### N95-23437

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### **Three-Dimensional Viscous Flow Analysis Inside a Turbine Volute**

C.Hah, J. Loellbach, and D. A. Greenwald NASA Lewis Research Center

L. Griffin and J. Ruf NASA Marshall Space Flight Center

A three-dimensional numerical method has been developed to analyze the complex flow field inside a turbine volute. Comparisons are made between solutions with different boundary conditions.

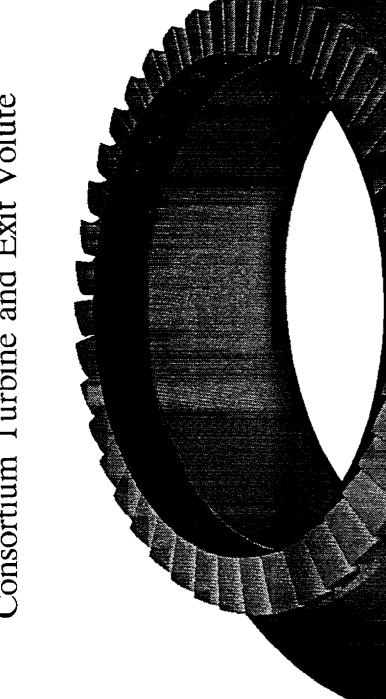
### Three-Dimensional Viscous Flow Analysis Inside a Turbine Volute

C. Hah, O. Kwon, J. Loellbach, and D. A. Greenwald

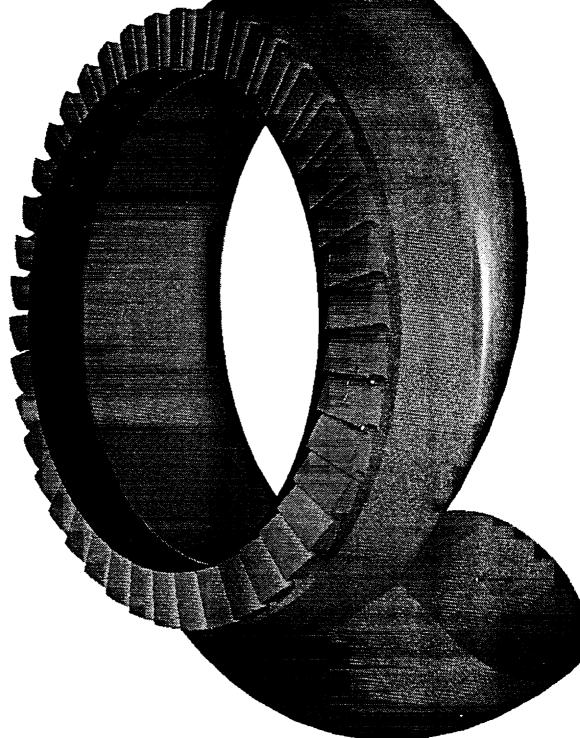
NASA Lewis Research Center

L. Griffin and J. Ruf NASA Marshall Space Flight Center

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Consortium Turbine and Exit Volute



### Objective

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o 3-D Turbulent Flow inside Volute

o Integrated Analysis of Turbine Stage and Inlet & Exit Volutes

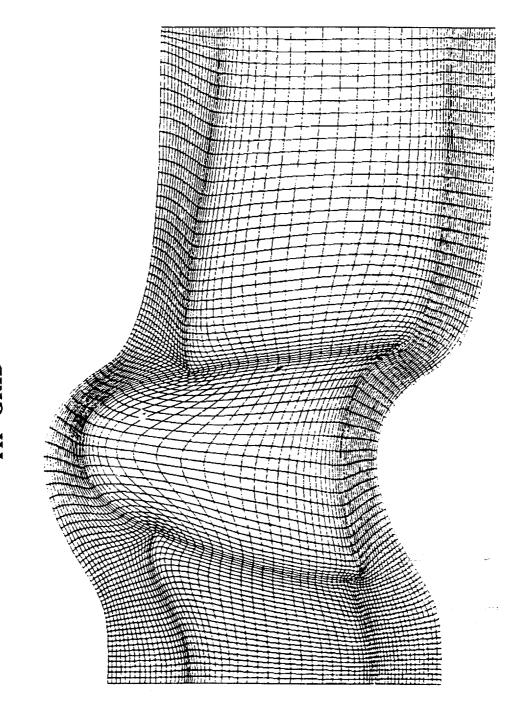
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### Approach

o 3-D Navier-Stokes Code (Structured Grid)

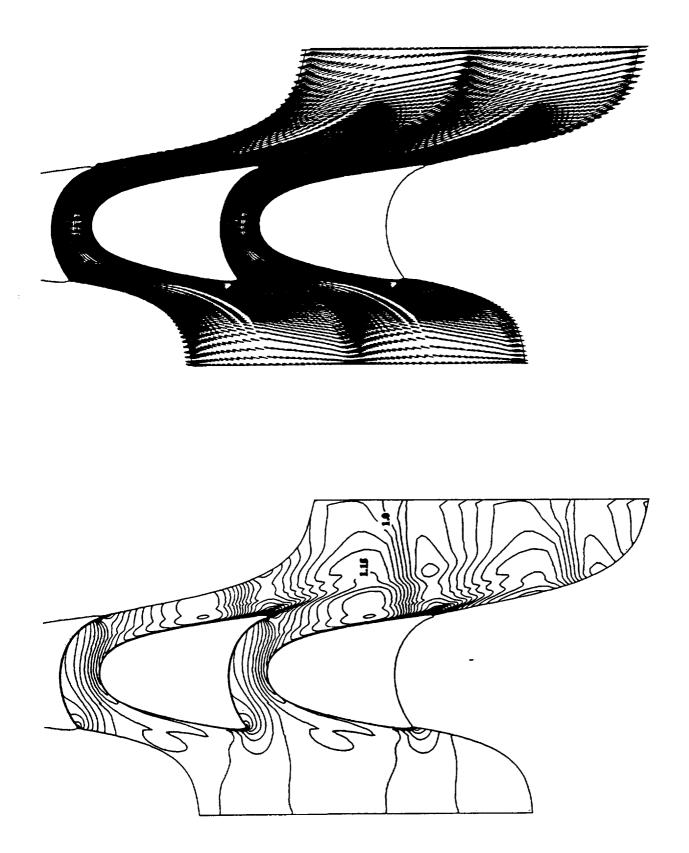
o 3-D Euler Code (Unstructured Grid)



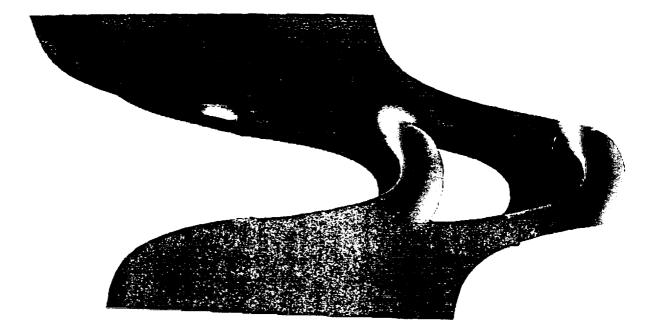
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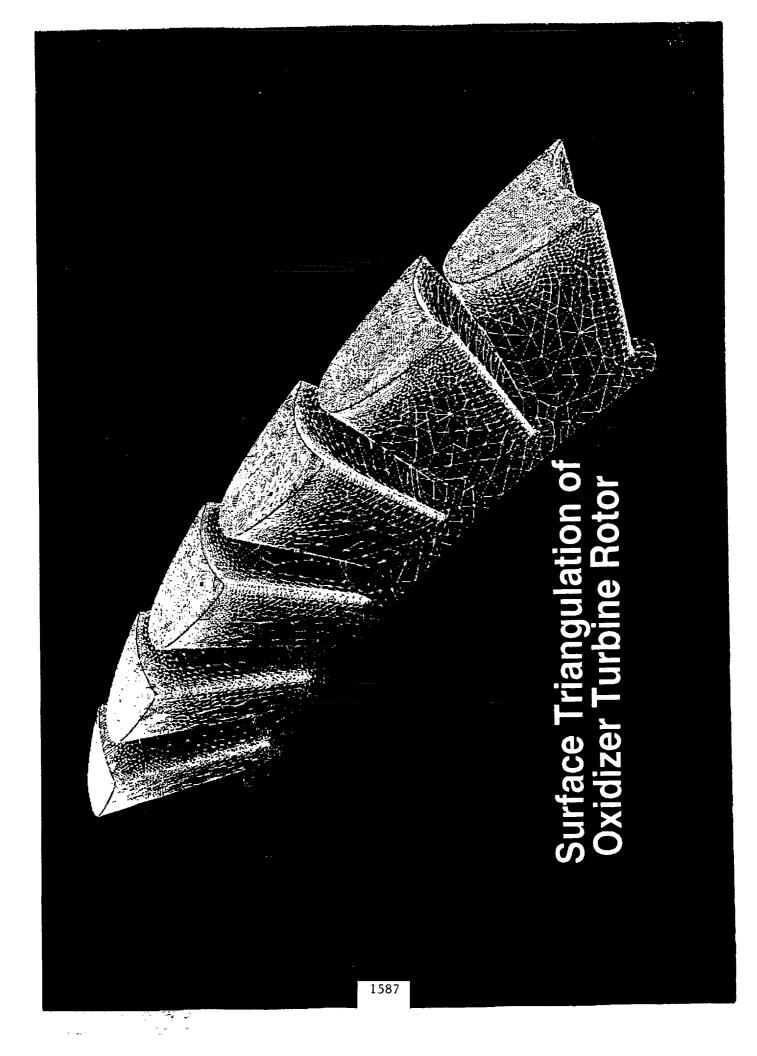
TIP GRID

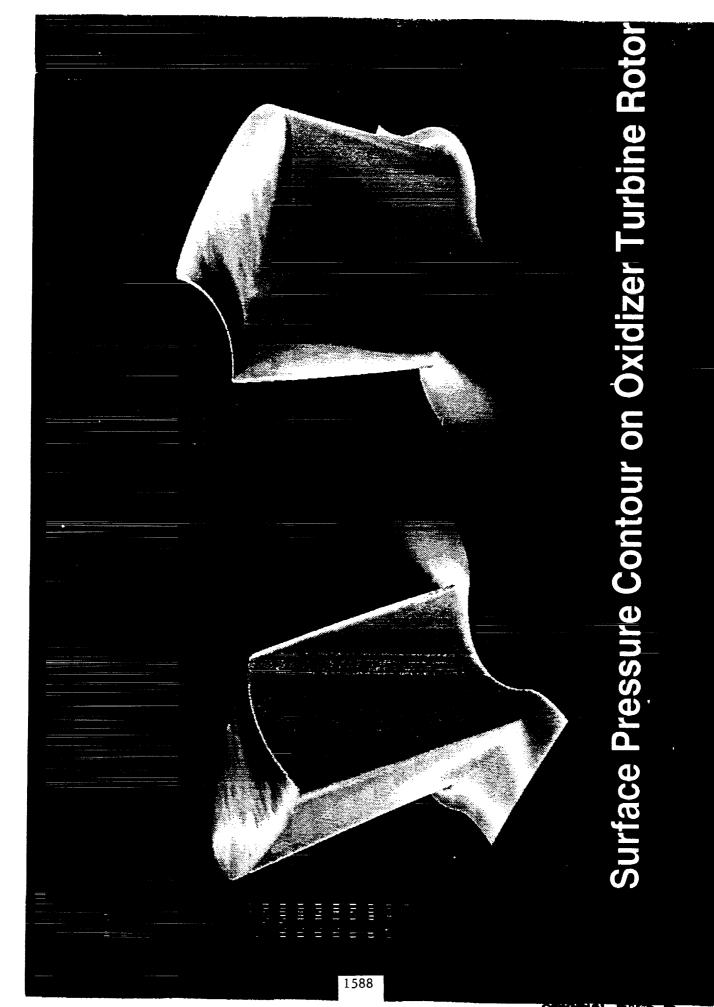
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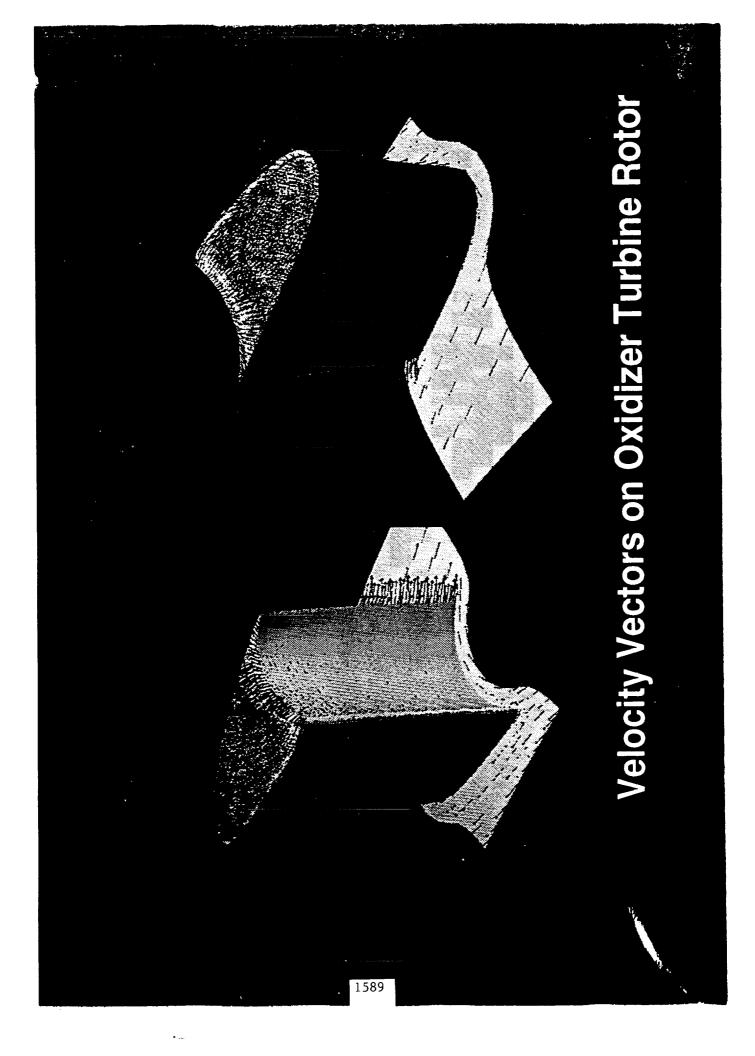


# MACH-NUMBER AND VELOCITY VECTORS (MID-SPAN)







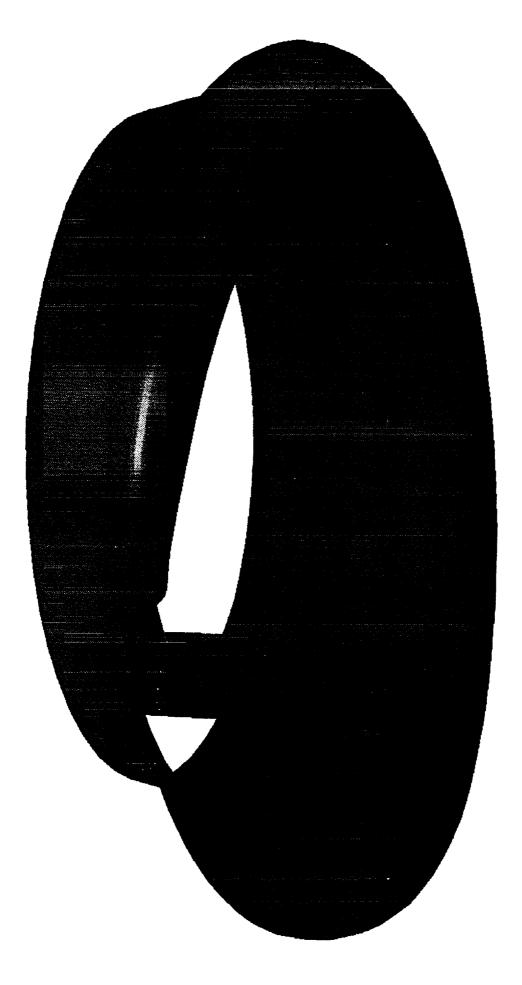




### Computational grid

# o Structured Grid (I-Grid): 51x32x46

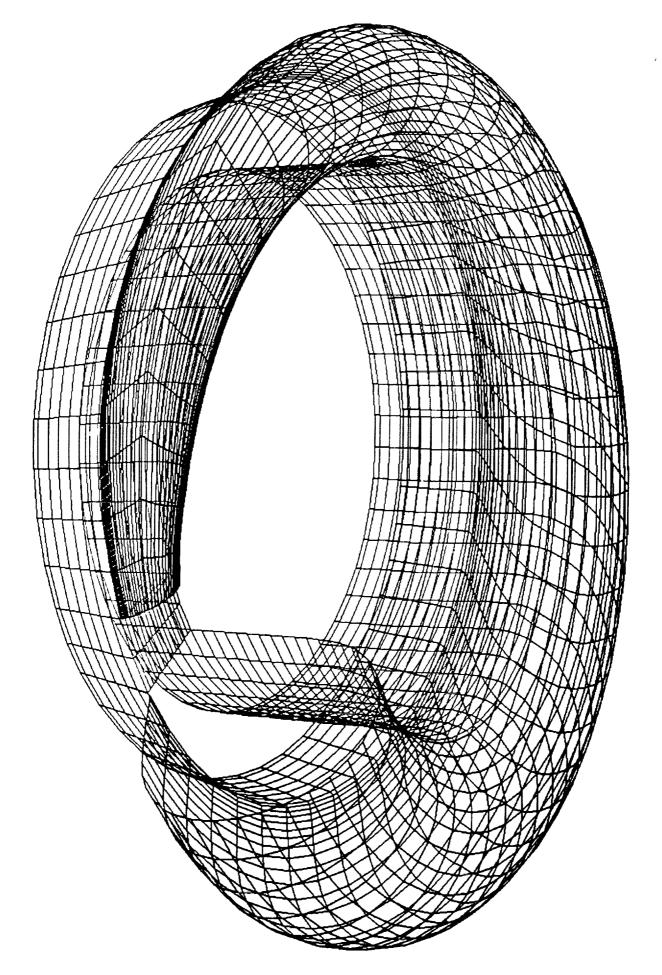
## o Unstructured Grid : 90,000 Nodes



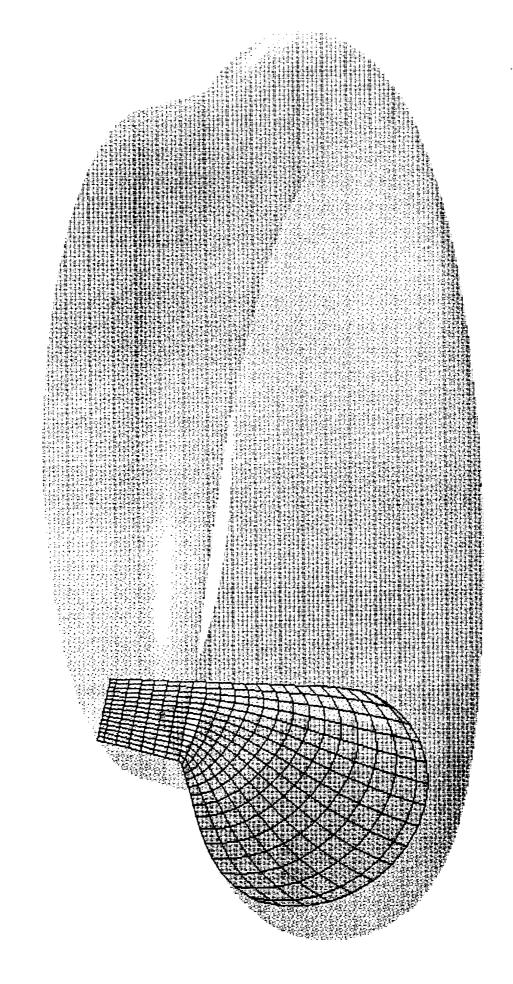
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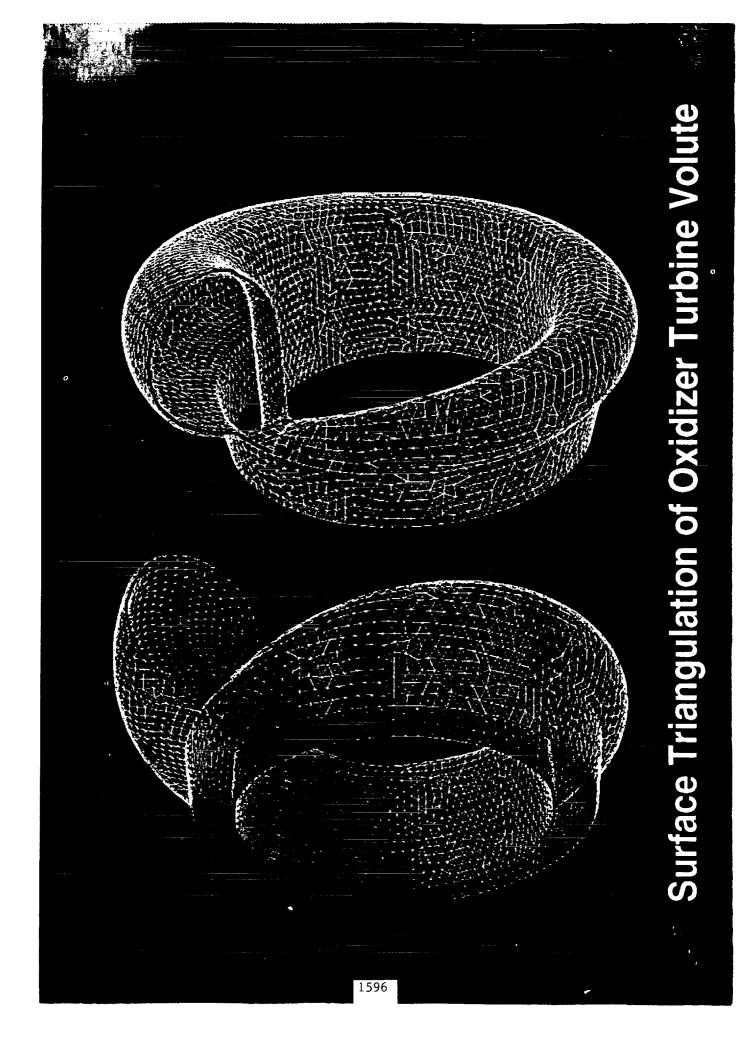
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### **BOUNDARY CONDITIONS**

**o** Inlet

- o Total Pressureo Total Temperatureo Two Velocity Components

### o Exit

o Static Pressure

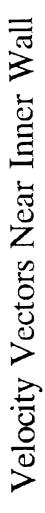
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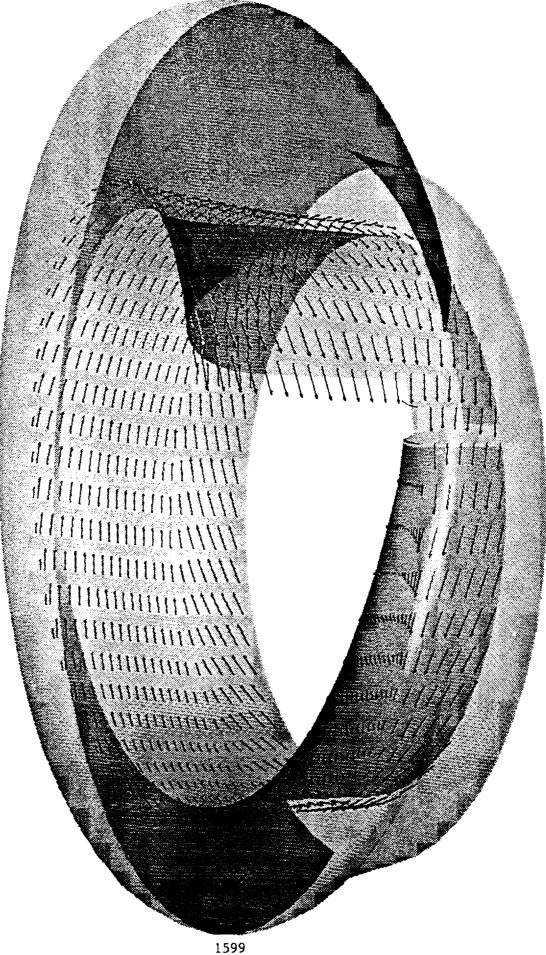
### HAH3D Viscous Flow Code

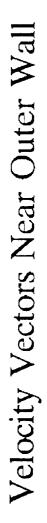
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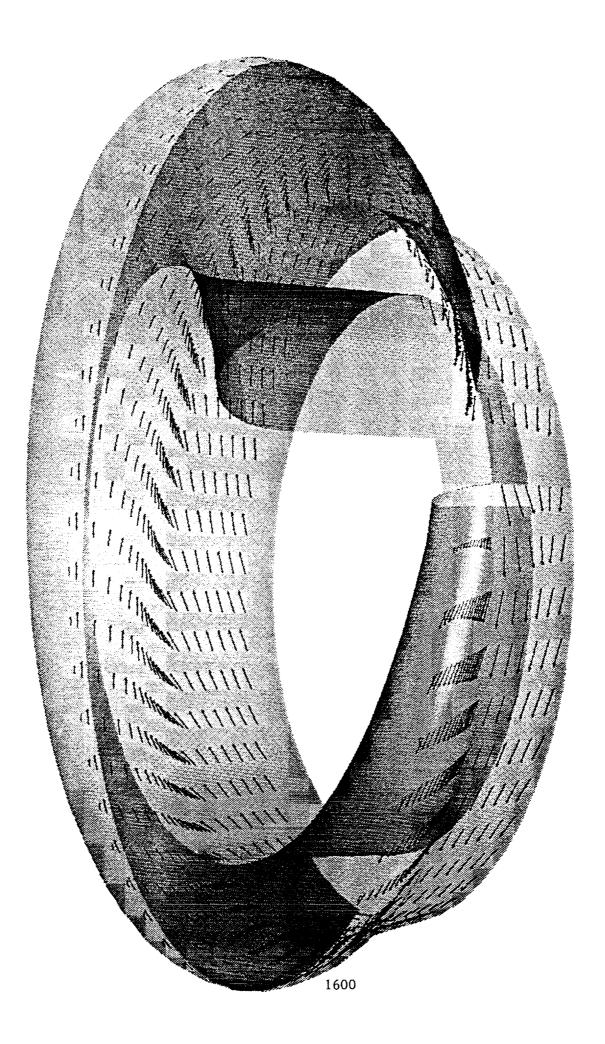
- o Three-Dimensional Navier-Stokes
- o Pressure-Based Control Volume
- o Two-Equation Turbulence Model with Low-Reynolds Number Extension
- o Incompressible, Transonic, Supersonic Flows

o Tested for Wide Range of Flows

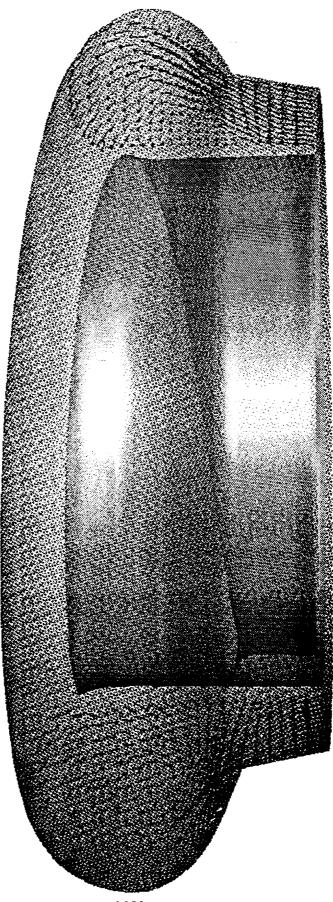






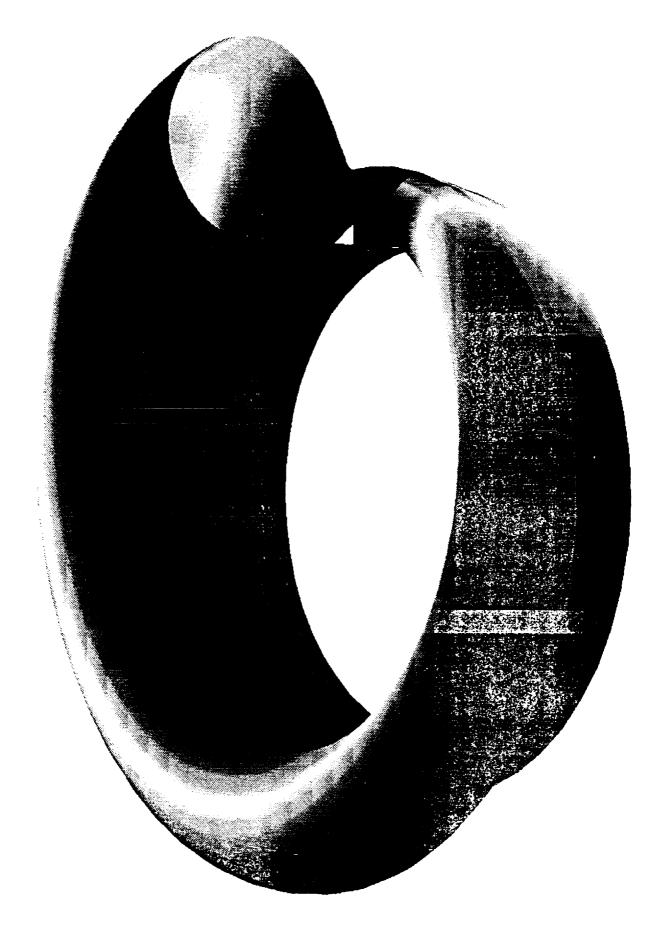


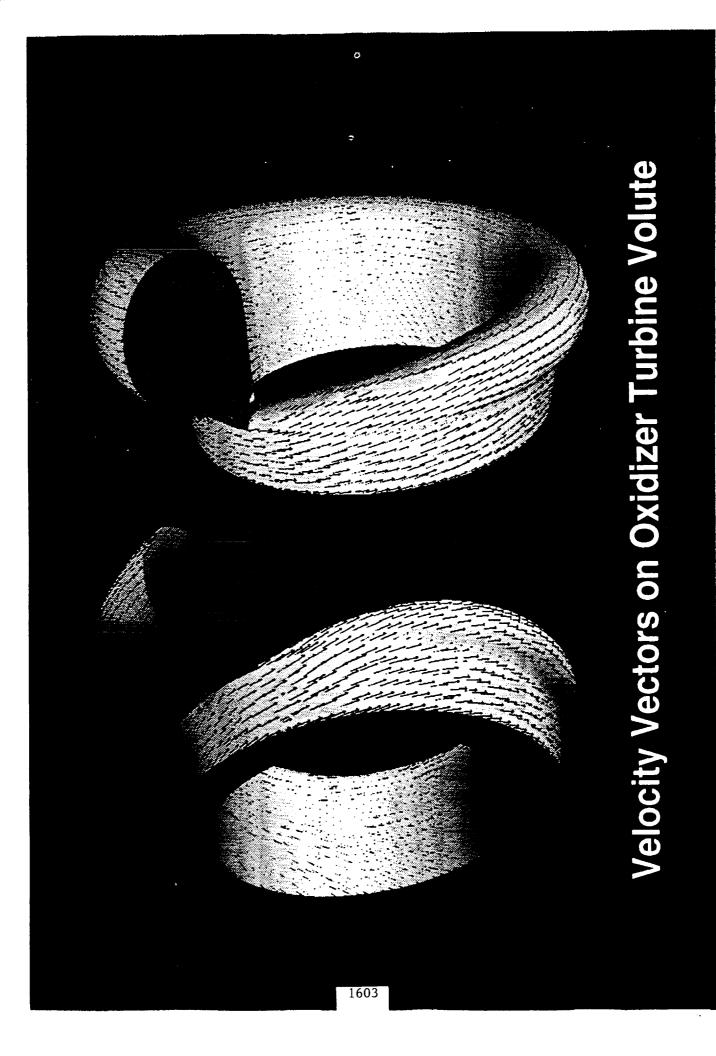
Crossflow Velocity Vectors

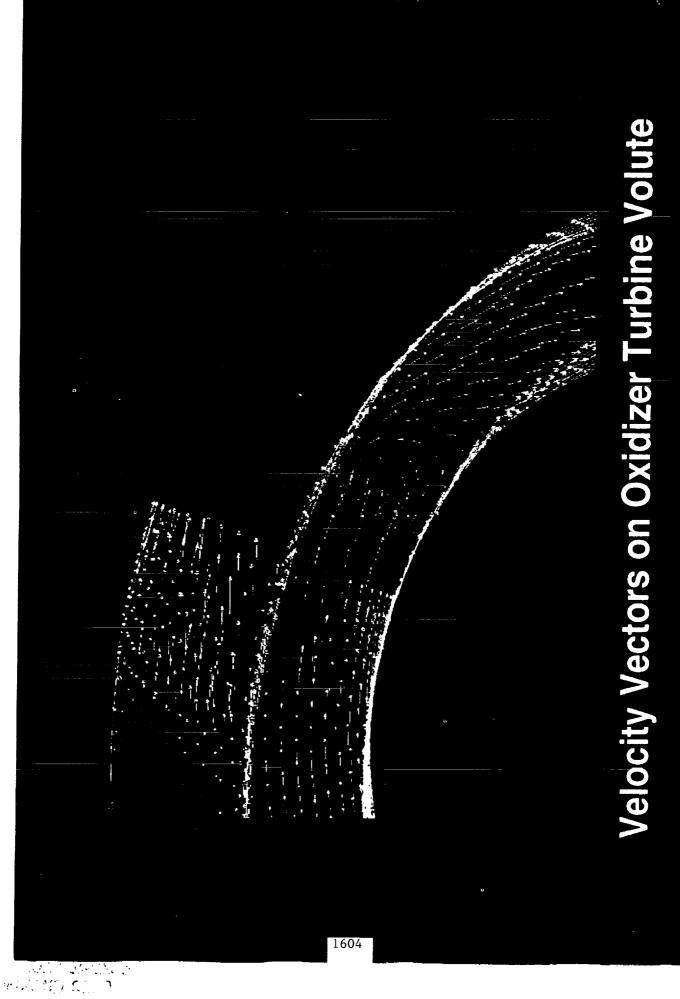




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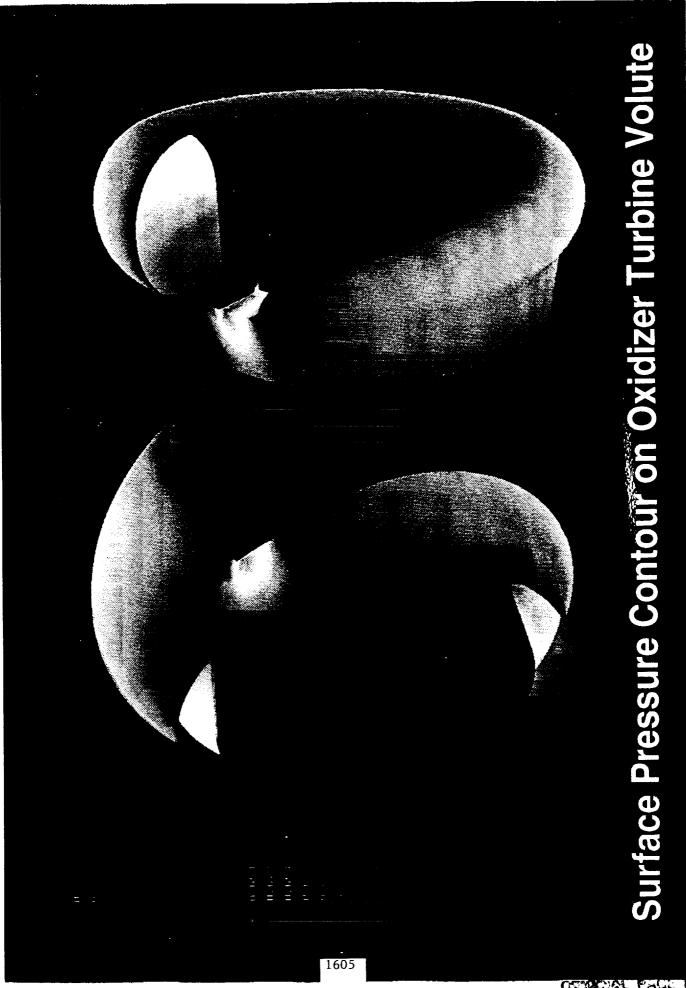






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### **Observation and Future Study**

- o Successful 3-D, N-S & Euler Flow Study for Volute Design
- o N-S & Euler Predict Different Flow features
- o Comparative Study between Structured and Unstructured Grid Methods
- o Further Verification Necessary

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