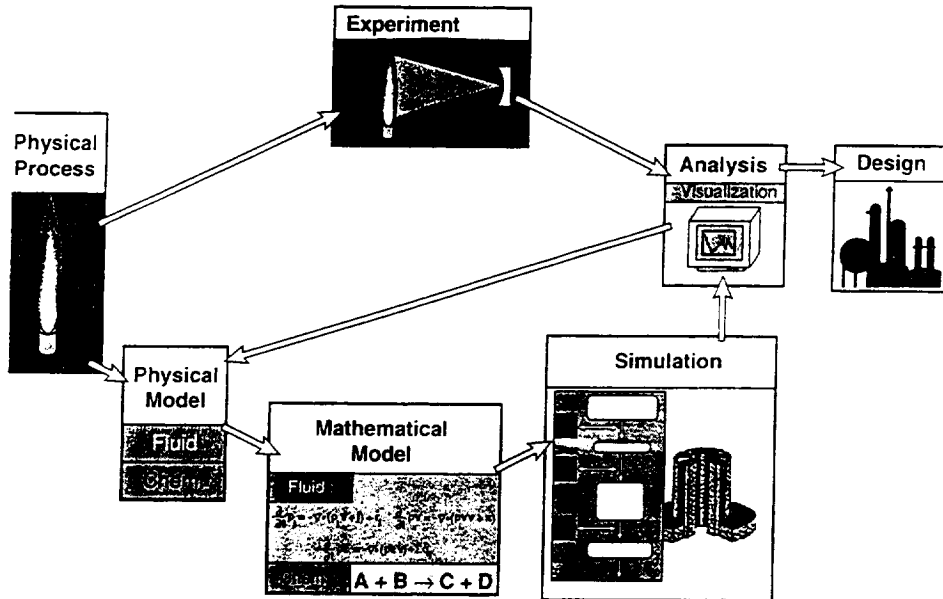


1995121478

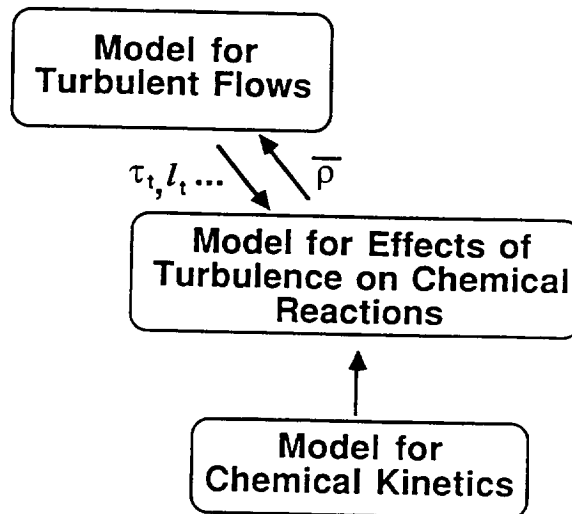
MODELING OF TURBULENT CHEMICAL REACTION

N95-27899

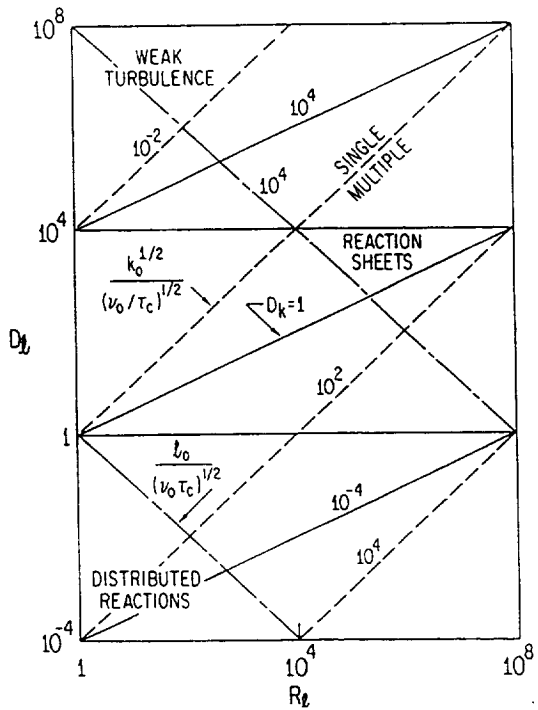
J.-Y. Chen  
Department of Mechanical Engineering  
University of California, Berkeley  
Berkeley, California



Modeling Turbulent Reacting Flows

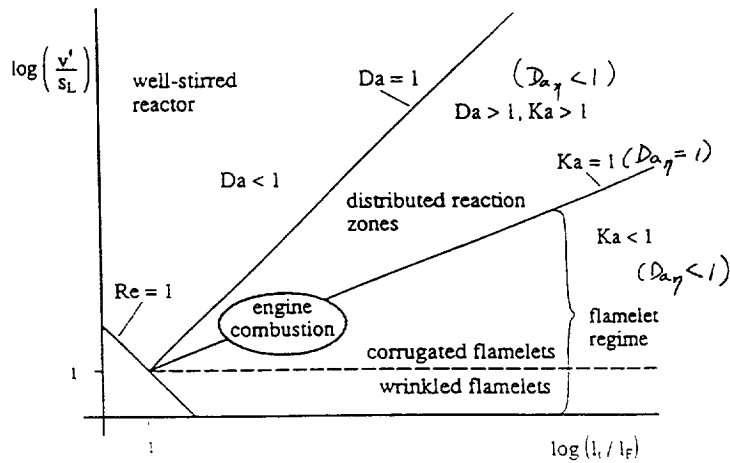


## Regimes of Turbulent Combustion



Turbulent Reactive Flows edited by P. A. Libby and F. A. Williams (1994)

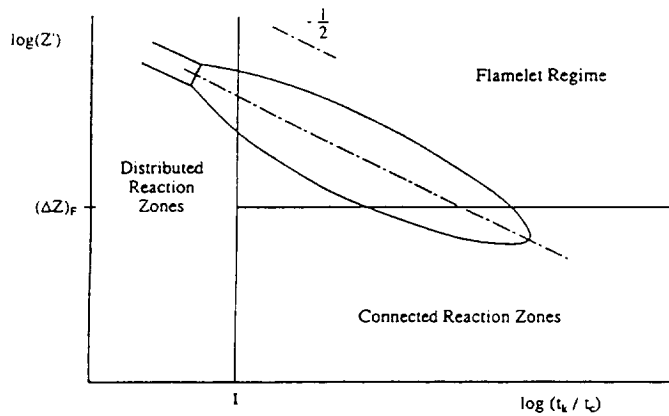
## Regimes of Premixed Turbulent Combustion



$$Ka = 1 / Da_\gamma$$

Turbulent Reactive Flows edited by P. A. Libby and F. A. Williams (1994)

## Regimes of Non-Premixed Turbulent Combustion



Turbulent Reactive Flows edited by P.A. Libby and F.A. Williams (1994)

### Chemical Closure Models

**(1) Laminar Chemistry**

$$\langle \rho w_i \rangle = \rho w_i(\bar{Y}_i, \bar{T})$$

**(2) Fast Chemistry**

$$\langle \rho w_i \rangle \approx -\frac{1}{2} \bar{\rho} \tilde{\chi}_f \frac{\partial^2 Y^c(f)}{\partial^2 f}$$

**(3) Flamelet model**

$$\langle \rho w_i \rangle = \iint \rho w_i(\eta, \chi_f) P_{f, \chi_f}(\eta, \epsilon_f) d\eta d\epsilon_f$$

**(4) Assumed PDF:**

$$\langle \rho w_i \rangle = \int \dots \int \rho w_i(\phi_i) \cdot P_\phi d\phi_1 d\phi_2 \dots d\phi_n$$

Assumed the shape of  $P_\phi$ .

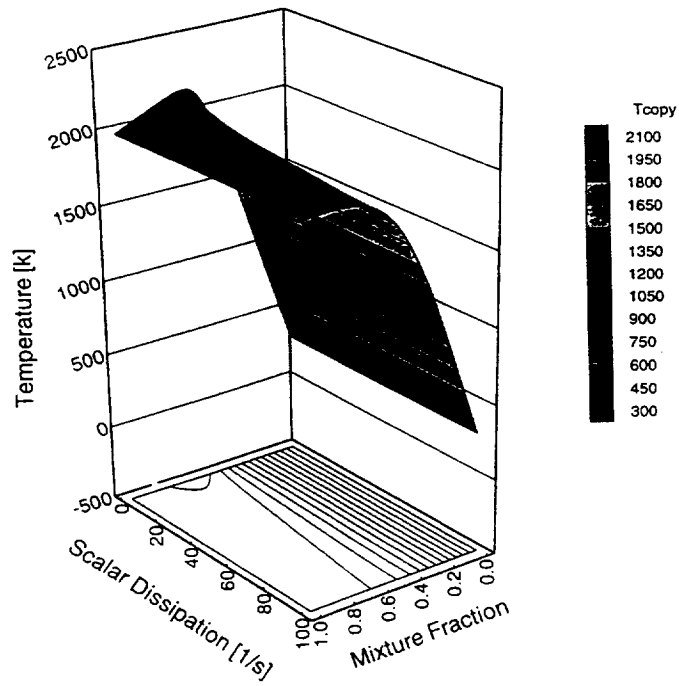
**(5) Scalar PDF method:**

Solve for  $P_\phi$  directly.

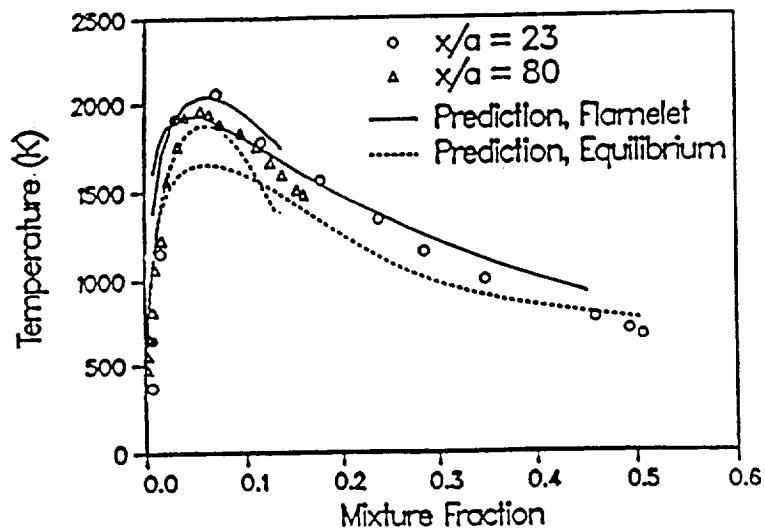
**(6) Conditional Moment Closure (CMC)**

$$\langle \rho w_i \rangle = \int \langle \rho w_i | \eta \rangle \cdot P_f(\eta) d\eta$$

Flamelet library with one side being burned premixed flame  $\phi=1.4$

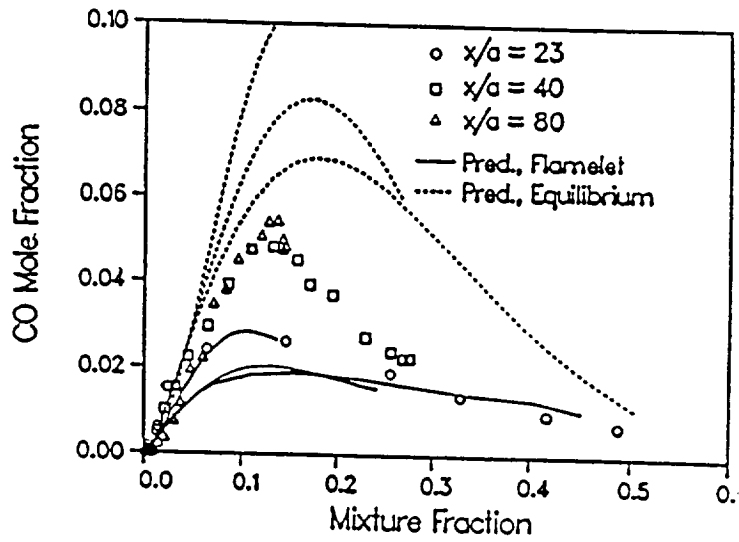


Flamelet Model: 69%H<sub>2</sub>+31%CH<sub>4</sub>  
Turbulent Jet Flame, Re<sub>y</sub>=10,000



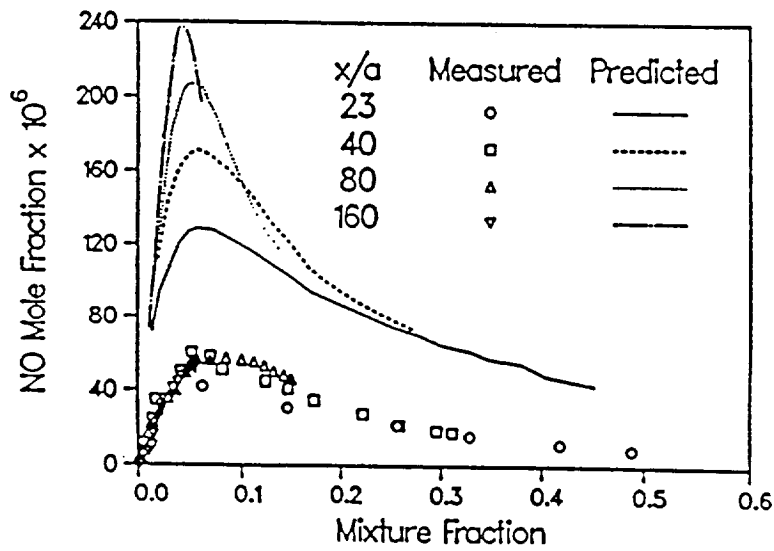
Vranos, et al. "Nitric Oxide Formation and Differential Diffusion in a Turbulent Methane-Hydrogen Diffusion Flame," 24th Symposium (International) on Combustion/The Combustion Institute, 1992/pp. 377-384

Flamelet Model: 69%H<sub>2</sub>+31%CH<sub>4</sub>  
Turbulent Jet Flame, Rev.=10,000



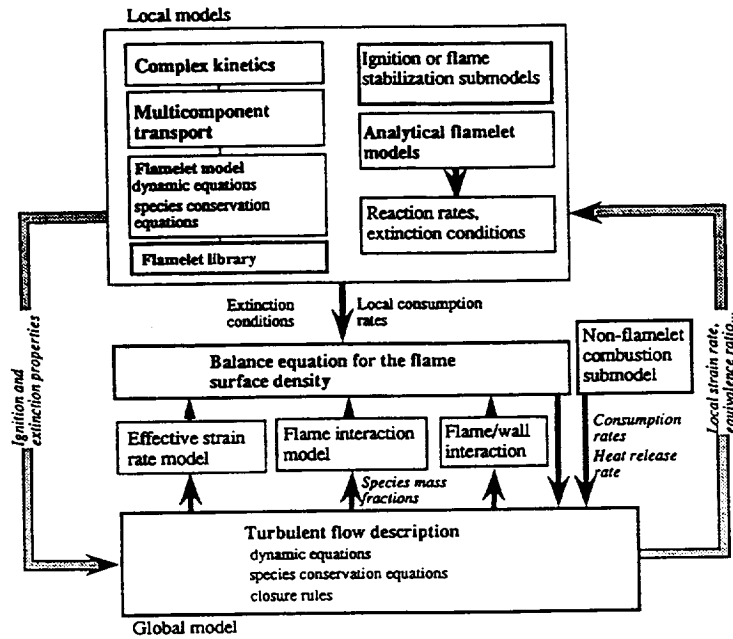
Vranos, et al. "Nitric Oxide Formation and Differential Diffusion in a Turbulent Methane-Hydrogen Diffusion Flame," 24th Symposium(International) on Combustion/The Combustion Institute, 1992/pp. 377-384

Flamelet Model: 69%H<sub>2</sub>+31%CH<sub>4</sub>  
Turbulent Jet Flame, Rev.=10,000



Vranos, et al. "Nitric Oxide Formation and Differential Diffusion in a Turbulent Methane-Hydrogen Diffusion Flame," 24th Symposium(International) on Combustion/The Combustion Institute, 1992/pp. 377-384

## Advanced Flamelet Approach



## Conditional Moment Closure (CMC)

Definition:

$$\langle Y_i | \eta \rangle \equiv \langle Y(\bar{x}, t) | f(\bar{x}, t) = \eta \rangle$$

Equation:

$$\langle \rho | \eta \rangle \frac{\partial \langle Y_i | \eta \rangle}{\partial t} + \langle \rho \tilde{u} | \eta \rangle \cdot \nabla \langle Y_i | \eta \rangle + \frac{\nabla \cdot \{ \langle \rho u' y' | \eta \rangle P_r(\eta) \}}{P_r(\eta)}$$

$$= \langle \rho w_i | \eta \rangle + \langle \rho D_i \nabla f \cdot \nabla f | \eta \rangle \frac{\partial^2 \langle Y_i | \eta \rangle}{\partial \eta^2}$$

Modeling:

$$\langle w_i | \eta \rangle \approx w_i(\langle T | \eta \rangle, \langle Y_i | \eta \rangle, \dots)$$

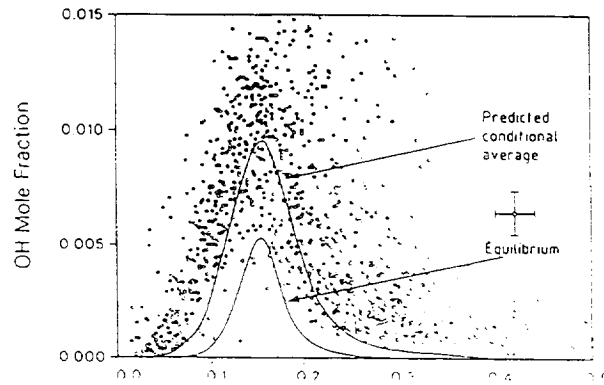
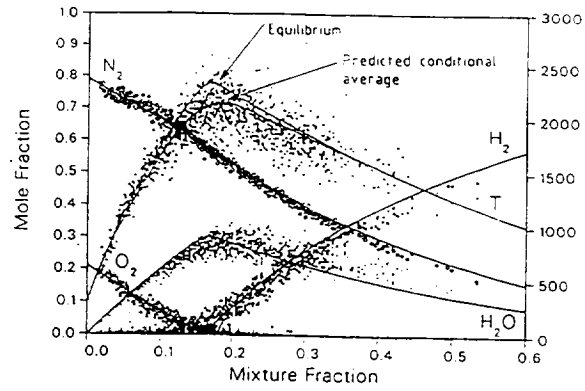
$$\langle \rho D_i \nabla f \cdot \nabla f | \eta \rangle \approx \langle \rho D_i \nabla f \cdot \nabla f \rangle \approx \frac{1}{2} \bar{\rho} \chi_i$$

$$\langle \rho \tilde{u} | \eta \rangle \approx \bar{\rho} \tilde{u}$$

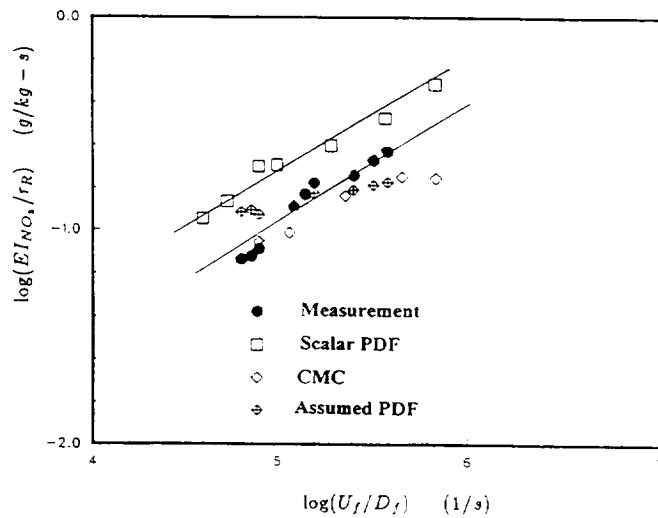
$$\langle \rho u' y' | \eta \rangle \approx 0$$

$$\langle \rho | \eta \rangle \approx \rho(\langle Y_i | \eta \rangle, \langle T | \eta \rangle)$$

## Conditional Moment Closure (CMC)



## NO<sub>x</sub> Emissions from Turbulent H<sub>2</sub> Jet Flames



## **Conditional Moment Closure (CMC)**

### **Applications:**

- **Incorporated into existing moment closure CFD codes for complex geometry flows**
- **Realistic Chemistry - Detailed or reduced**

### **Research issues:**

- **Modeling of conditional statistics**
- **Preferential diffusion**
- **Parallel computing algorithms**

## **Probability Density Function (PDF)**

### **Applications:**

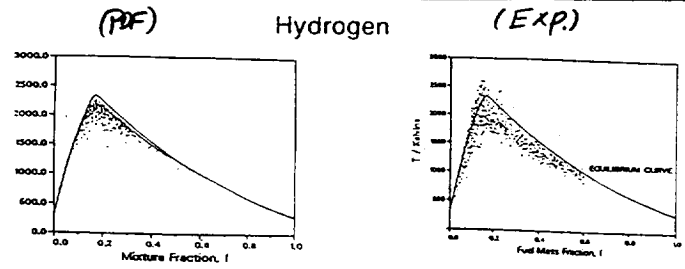
- **NO<sub>x</sub> from methane jet flames with reduced chemistry**
- **Sooting flames**
- **2-D flows**

### **Research Topics:**

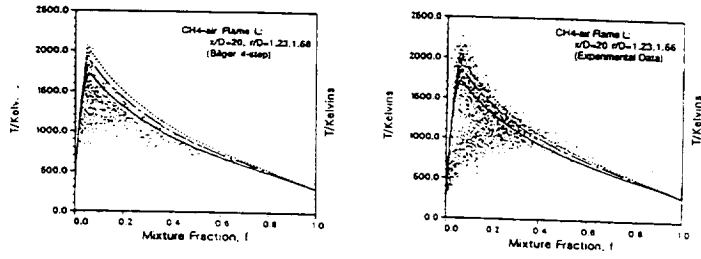
- **Mixing model**
- **Extension to droplet spray & particle laden flows**
- **Preferential diffusion**
- **Efficient stochastic algorithm**
- **Construction of chemical tables**
- **Parallel computing - 3D Flows or 2D flows with complex chemistry**



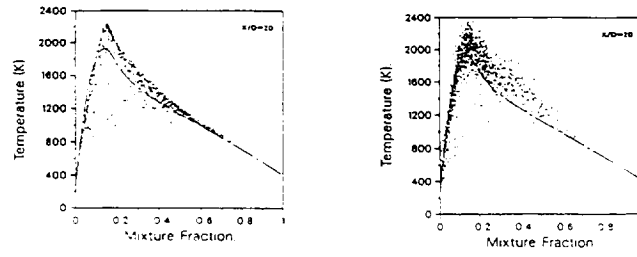
## Departures From Chemical Equilibrium



### Methane



### Methanol



## Mixing Models for PDF Methods

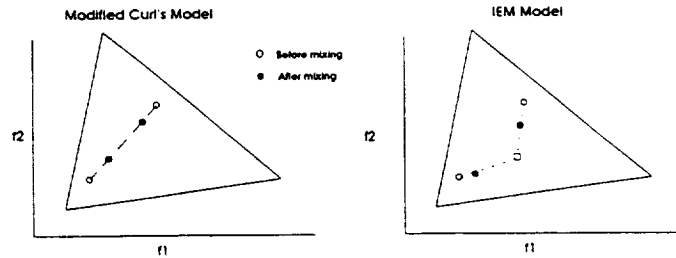
- Modified Curl's Model (stochastic)

$$-\sum_{\alpha=1, \beta=1}^k \frac{\partial^2}{\partial \psi_\alpha \partial \psi_\beta} \left\{ \left\langle \varepsilon_{\alpha\beta} | \bar{\phi} = \bar{\psi} \right\rangle \bar{P}_\phi(\bar{\psi}, t) \right\} = \frac{1}{\tau_{mix}} \left\{ \iint_{\psi' \psi''} \left[ \bar{P}_\phi(\psi', t) \bar{P}_\phi(\psi'', t) H(\psi', \psi'' | \bar{\psi}) - \bar{P}_\phi(\bar{\psi}, t) \right] d\psi' d\psi'' \right\}$$

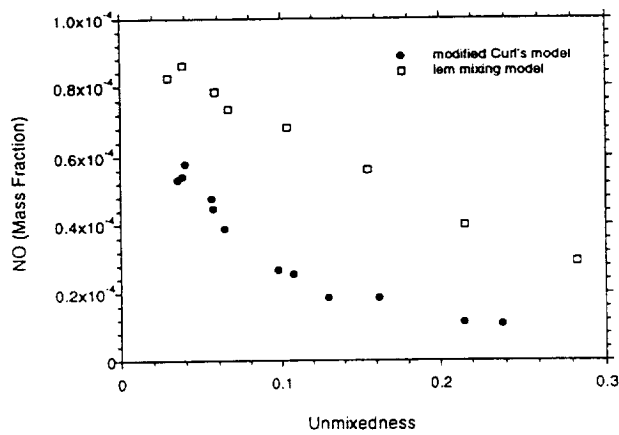
- IEM (Interaction-by-Exchange-with-the-Mean) Model (deterministic)

$$-\sum_{\alpha=1, \beta=1}^k \frac{\partial^2}{\partial \psi_\alpha \partial \psi_\beta} \left\{ \left\langle \varepsilon_{\alpha\beta} | \bar{\phi} = \bar{\psi} \right\rangle \bar{P}_\phi(\bar{\psi}, t) \right\} = \frac{C_\phi}{2\tau_{mix}} \frac{\partial}{\partial \psi_\alpha} \left[ (\bar{\psi} - \bar{\phi}) \bar{P}_\phi(\bar{\psi}, t) \right]$$

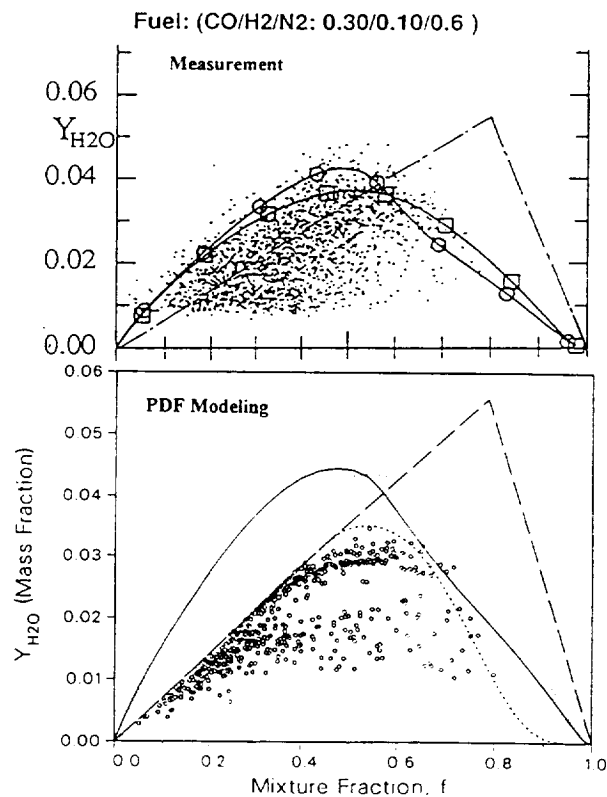
Mixing Frequency:  $\omega_{mix} = 1/\tau_{mix}$



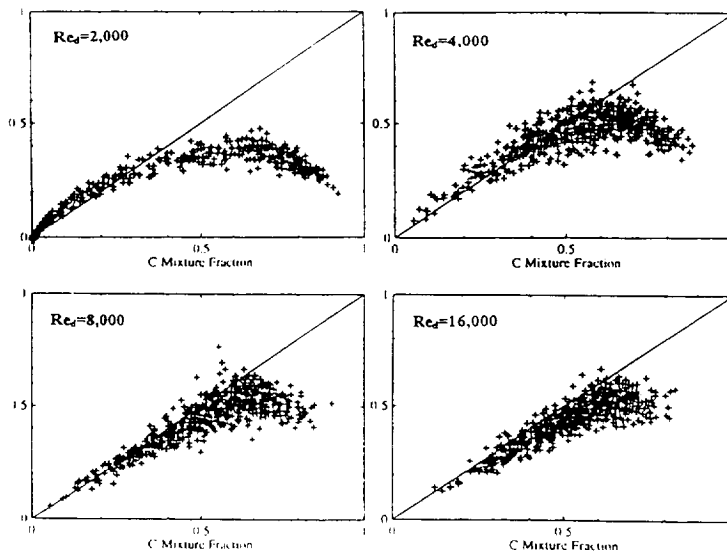
### PaSR: H2/NOx Detailed Chemistry $\phi=1$ $\tau=1$ ms



## Comparison of Predicted and Measured H<sub>2</sub>O Mass Fractions Turbulent Nonpremixed Jet Flames

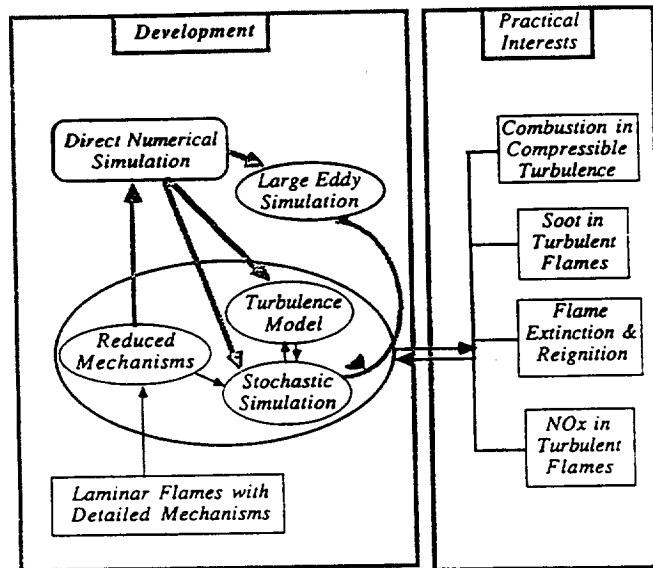


### Experimental Evidence of Preferential Diffusion in Turbulent Jet Flames (Fuel: 36% H<sub>2</sub> + 64% CO<sub>2</sub>)



"Differential Molecular Diffusion in Reacting and Nonreacting Turbulent Jets of H<sub>2</sub>/CO<sub>2</sub> mixing with Air," L.L. Smith Ph.D. Thesis, University of California at Berkeley (1994)

## Computation of Turbulent Reacting Flows



## INTRODUCTION TO TURBULENCE SUBPROGRAM

T.-H. Shih and J. Zhu  
Institute for Computational Mechanics in Propulsion  
and Center for Modeling of Turbulence and Transition  
NASA Lewis Research Center  
Cleveland, Ohio

### OBJECTIVES

- A means for CMOTT to interact with industry
- A vehicle for technology transfer to industry

### CONCEPT OF TURBULENCE MODULE

- Exact CFD equations:

$$\frac{D\rho U_i}{Dt} = \frac{\partial}{\partial x_j} \left[ \mu \left( \frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} - \frac{2}{3} \frac{\partial U_k}{\partial x_k} \delta_{ij} \right) - \rho \overline{u_i u_j} \right] - \frac{\partial P}{\partial x_i}$$

- Reynolds stresses will be recasted as:

$$-\rho \overline{u_i u_j} \equiv \underbrace{\mu_T \left( \frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} - \frac{2}{3} \frac{\partial U_k}{\partial x_k} \delta_{ij} \right)}_{T_{ij}} + \underbrace{[-\rho \overline{u_i u_j} - \mu_T \left( \frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} - \frac{2}{3} \frac{\partial U_k}{\partial x_k} \delta_{ij} \right)]}_{\mu_T \equiv C_\mu \frac{k^2}{\varepsilon}}$$

- CFD equations become:

$$\frac{D\rho U_i}{Dt} = \frac{\partial}{\partial x_j} \left[ (\mu + \mu_T) \left( \frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} - \frac{2}{3} \frac{\partial U_k}{\partial x_k} \delta_{ij} \right) \right] + \frac{\partial T_{ij}}{\partial x_j} - \frac{\partial P}{\partial x_i}$$

- The task of turbulence module: Provide  $\mu_T$  and  $T_{ij}$

- Turbulence Module:

- ◇ Input:  $U_i, \rho$  and  $\mu \dots$  from the mean flow solver

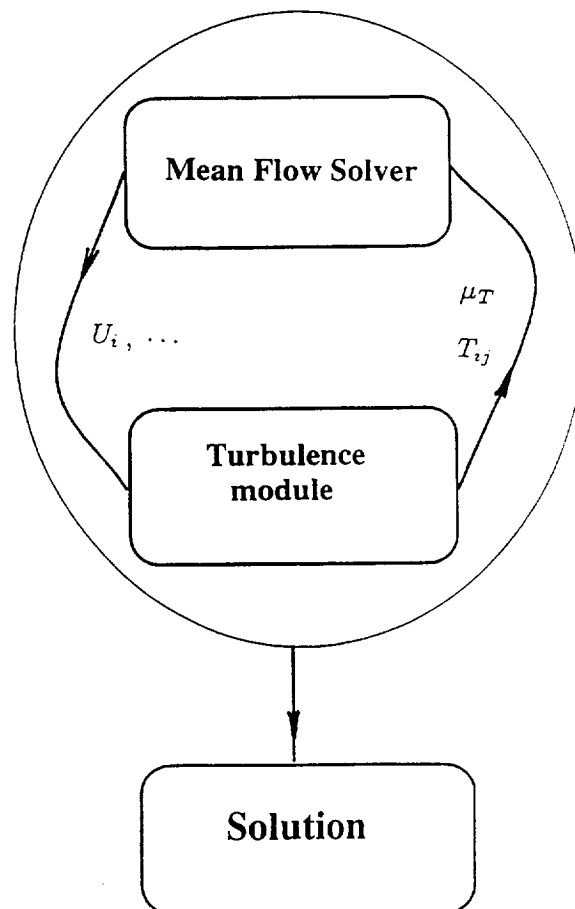
- ◇ Output:

$$\mu_T = C_\mu \frac{k^2}{\varepsilon} \quad \left[ \frac{Dk}{Dt} = \dots, \quad \frac{D\varepsilon}{Dt} = \dots \right]$$

$$T_{ij} = -\rho \overline{u_i u_j} - \mu_T \left( \frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} - \frac{2}{3} \frac{\partial U_k}{\partial x_k} \delta_{ij} \right)$$

- ◇ Models for  $\rho \overline{u_i u_j}$

- One- and two-equation eddy viscosity models
    - Reynolds stress algebraic equation models
    - Reynolds stress transport equation models



## Module with CMOTT research code (incompressible)

- CFD equations in CMOTT research code:

$$\frac{D\rho U_i}{Dt} = \frac{\partial}{\partial x_j} [(\mu + \mu_T) \left( \frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right)] + \frac{\partial}{\partial x_j} T_{ij} - \frac{\partial P}{\partial x_i}$$

- Turbulence module: provide  $\mu_T$  and  $T_{ij}$ 
  - ◇ Built-in models without wall function:
    - Launder-Sharma and Chien  $k - \epsilon$  models
    - CMOTT  $k - \epsilon$  model
  - ◇ Built-in models with wall function:
    - $k - \omega$  model, standard  $k - \epsilon$  model
    - CMOTT  $k - \epsilon$  model
    - CMOTT Reynolds stress algebraic equation model

## Module with NPARC code

- CFD equations in NPARC code:

$$\frac{D\rho U_i}{Dt} = \frac{\partial}{\partial x_j} [(\mu + \mu_T) \left( \frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} - \frac{2}{3} \frac{\partial U_k}{\partial x_k} \delta_{ij} \right)] - \frac{\partial P}{\partial x_i}$$

- Turbulence module (present time): provide isotropic  $\mu_T$ 
  - ◇ Build-in models without wall function:
    - Baldwin-Lomax model and Chien  $k - \epsilon$  model
    - CMOTT  $k - \epsilon$  model
  - ◇ Further development:
    - Models with wall function
    - Reynolds stress algebraic equation models
    - Reynolds stress transport equation models

## **Joint Program with Industry on Turbulence Module**

- For those who want to use the available modules:
  - ◊ Need interface program for particular industry codes
    - Grid informations, Boundary treatment, ...
- For those who want a module for their own codes:
  - ◊ Need modules exclusively for particular industry codes
- Maintain and update the turbulence modules along with model development.



## DESCRIPTION OF TURBULENCE SUB-PROGRAM

J. Zhu

Institute for Computational Mechanics in Propulsion  
NASA Lewis Research Center  
Cleveland, Ohio

### General Transport Equations

$$\frac{\partial}{\partial t}(rJ^{-1}\rho\phi) + \frac{\partial}{\partial \xi_i}(C_i\phi - D_i\phi) = rJ^{-1}S_\phi$$

- Non-dimensional form ( $\mu, \mu_t \Leftrightarrow \mu/Re, \mu_t/Re$ )
- Conservative form
- Cartesian velocity components
  1. Easy to transform (chain rule)
  2. No curvature terms

## Discretization

- Finite-volume method
- Source term
 
$$S_\phi = S_1 + S_2\phi, \quad S_1 \geq 0 \text{ and } S_2 \leq 0$$
- Transient term
  1. 1st-order fully implicit scheme
  2. 2nd-order three-level fully implicit scheme
- Diffusion term
 

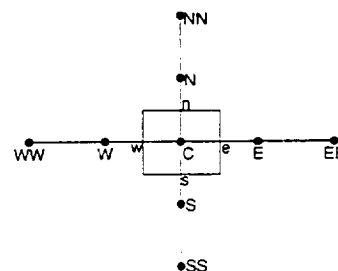
Standard central differencing scheme

- Convection term: HLLPA scheme  
(Hybrid Linear/Parabolic Approximation)

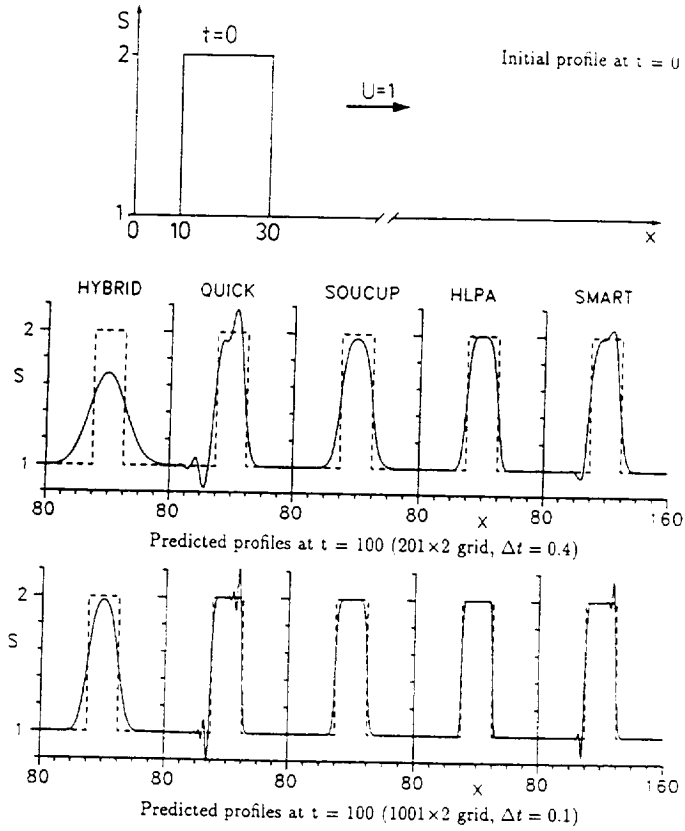
$$\phi_w = \phi_W + \gamma(\phi_C - \phi_W)\hat{\phi}_W, \quad \hat{\phi}_W = \frac{\phi_W - \phi_{WW}}{\phi_C - \phi_{WW}}$$

$$\gamma = \begin{cases} 1 & \text{if } |\hat{\phi}_W - 0.5| < 0.5 \\ 0 & \text{otherwise} \end{cases}$$

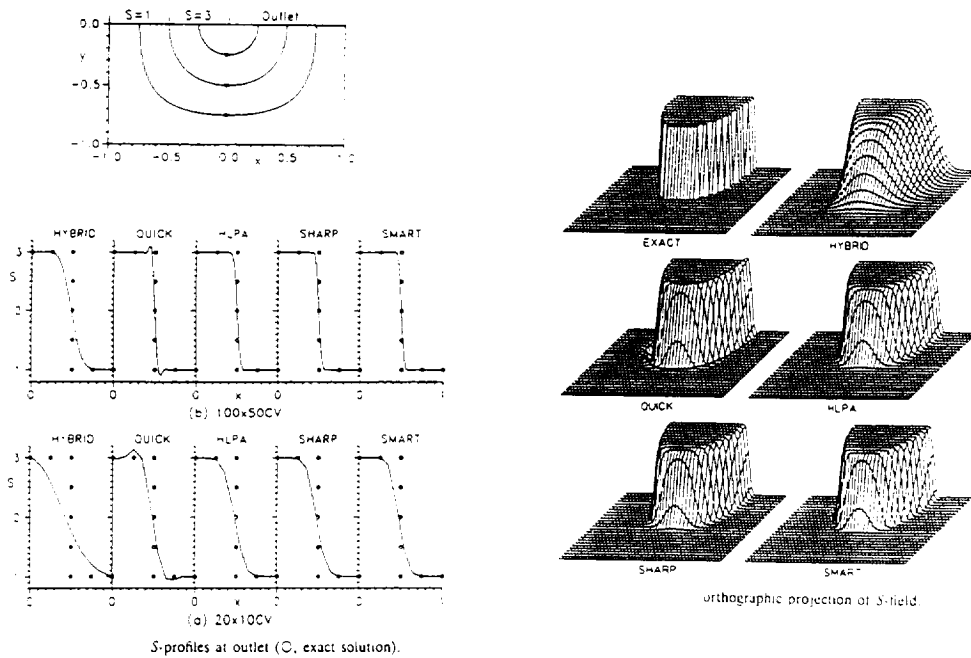
- Second-order accurate
- Bounded (non-oscillatory)
- Diagonally dominant matrix



# Example 1



# Example 2



S-profiles at outlet (O, exact solution).

## Solution Procedure

- Non-delta form

Positiveness ( $\phi \geq 0$  but  $\Delta\phi$  may  $< 0$ )

Simple linearization

- Algebraic equations

$$A_C\phi_C = A_W\phi_W + A_E\phi_E + A_S\phi_S + A_N\phi_N + S$$

$A$ 's,  $S \geq 0$

- Decoupled solution
- Alternating direction TDMA solver

## Boundary Conditions

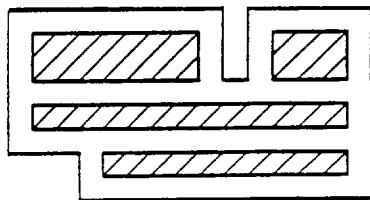
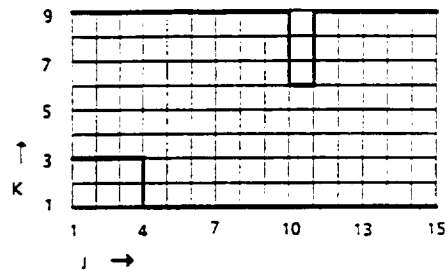
- Inflow:  $\phi$  specified
- Outflow: Fully-developed condition
- Symmetry:  $\partial\phi/\partial n = 0$
- Wall:
  1. Low-Reynolds number turbulence models
  2. Standard wall-function approach

## Sub-Programs

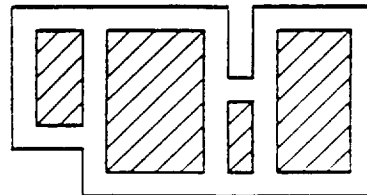
- NPARC2D version
  - Plane or axisymmetric, without swirling
  - Compressible
  - Non-vectorized
- FAST2D version
  - Plane or axisymmetric, with or without swirling
  - Incompressible
  - Vectorized

### NPARC2D Version

- Grid arrangement
  - Control volume centers
  - Boundary nodes
  - Embedded bodies



J-Patches

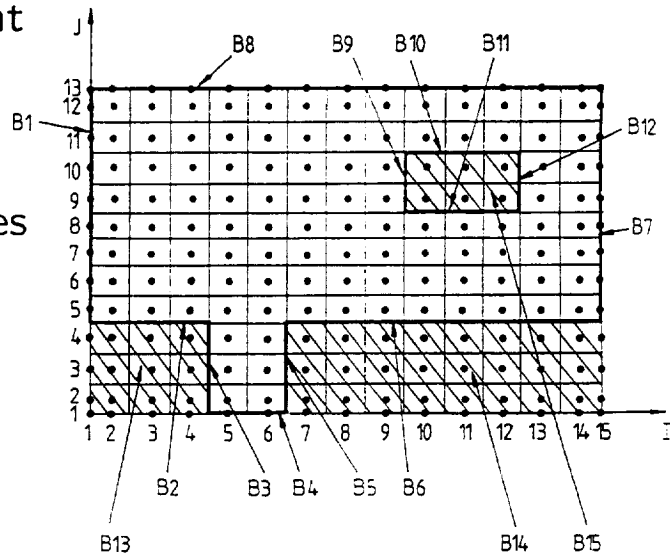


K-Patches

- Input from the main code
  1. Geometric quantities:  $x, y, \xi_x, \xi_y, \eta_x, \eta_y, J$
  2. Flow variables:  $\mu, J^{-1}\rho, J^{-1}\rho U, J^{-1}\rho V, J^{-1}E$
  3. Patch control:  $5 \times 2$  parameters
  4. Boundary conditions:  $7 \times 2$  parameters
  
- Output
  1. To the main code:  $\mu_t$
  2. For post-processing:  $K, \epsilon, y^+, y_n, f_\mu$

## FAST2D Version

- Grid arrangement
  - CV centers
  - Boundary nodes
  - Embedded bodies



- Vectorization

Single-index:

$$ii = i + (j-1)ni$$

$$\phi(i,j) = \phi(ii)$$

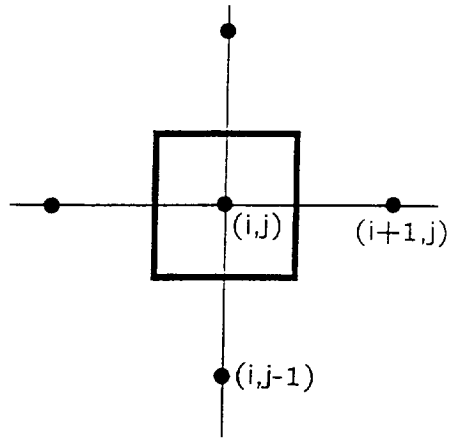
$$\phi(i+1,j) = \phi(ii+1)$$

$$\phi(i,j-1) = \phi(ii-ni)$$

Control parameter:

$$KBLK = \begin{cases} 1 & \text{for computational nodes} \\ 0 & \text{otherwise} \end{cases}$$

$$\phi = KBLK \cdot \phi_c + (1-KBLK) \phi_b$$



- Input from the main code

1. Geometric quantities:  $x, y, x_\xi, x_\eta, y_\xi, y_\eta, J$
2. Flow variables:  $\mu, \rho, U, V, W, C_w, C_s$
3. Vectorization parameters
4. Boundary parameters

- Output

1. To the main code:  $\mu_t, T_{ij}$
2. For post-processing:  $K, \epsilon, y^+, y_n, f_\mu$





# OVERVIEW OF PROBABILITY DENSITY FUNCTION (PDF) MODELING AT LeRC

D.R. Reddy  
Internal Fluid Mechanics Division  
NASA Lewis Research Center  
Cleveland, Ohio

## OBJECTIVE

Accurately model the effect of turbulence on  
chemical reactions in a fluid flow

## APPROACH

Use Probability Density Function (PDF) model -  
Express dependent variables as functions  
representing statistically realizable events

## POSSIBLE MODELING STRATEGIES

1. Evolution PDF - solve for function
  - a. Joint PDF for velocities and chemical species
  - b. Joint PDF for only chemical species  
& energy
  
2. Assumed PDF - function prescribed  
Limited range of applicability -  
reaction time  $\ll$  or  $\gg$  turbulence time scale

## **CURRENT APPROACH**

- **Develop evolution PDF model for compressible reacting flows & extend to spray combustion**
- **Solve for joint PDF for species and energy using Monte-Carlo technique**
- **Couple with conventional CFD codes**

## **AREAS OF IMPACT**

- **NO<sub>x</sub> Prediction - HSCT and AST application**
- **Spray combustion - swirling turb. reacting flows**
- **Scramjet flow path analysis**
- **Ignition kinetics - prediction of blow-off, etc.**
- **Combustion instability studies**

## CODE FEATURES

- **Modular - can be coupled with any CFD code**
- **Applicable for compressible flows with discontinuities**
- **Monte-Carlo solver for generalized curvilinear coordinate system**
- **Easily adaptable for parallel computation (currently under progress)**

## CURRENT STATUS

- **2-D and axisymmetric version released**  
(default H<sub>2</sub>-air chemistry - 5 species)  
- parallel version to be released
- **3-D version demonstrated for supersonic combustion (jet in cross flow)**  
- validation planned for HSCT-type configurations
- **General chemistry (CHEMKIN)**  
- Hydrocarbon spray combustion case currently under study
- **CFD codes used - RPLUS, ALLSPD, & SIMPLE-type**

## **FUTURE PLANS**

- **Further application/validation of 3-D model**
- **Improved closure models - mixing and turbulence  
(use available DNS data)**
- **Parallel processing - workstation clusters**
- **Unsteady applications - long-term**
- **Extend scope of impact**