

Bipolar Nickel-Metal Hydride Battery
Development Project

prepared by:

John H. Cole

presentation to:

NASA Aerospace Battery Workshop

Contract Agency:

NASA Lewis Research Center

October 27, 1998

Electro Energy Inc.

Shelter Rock Lane

Danbury, CT 06810

203-797-2699

1-800-BIPOLAR

203-797-2697 fax

516-44
615-263
372365

PROJECT OBJECTIVE: Design of 1 kW Bipolar Ni-MH Battery for LEO-Satellite

Sponsor: NASA Lewis Research Center

Contractor: Electro Energy Inc.

Subcontractor: Eagle-Picher Industries, Inc.
Design Automation Associates
Rhone-Poulenc, Inc.
Rutgers College of Engineering

Duration (Deliverable): 5 Year (Flight-weight Design Package)
Option (Deliverable): 18 Month (3 Flight Quality Batteries)

Technical Requirements:
a. 5-Year Operation in LEO Regime
(C:55 min, D: 35 min)
b. 40% DOD
c. 28 V @ 1kW

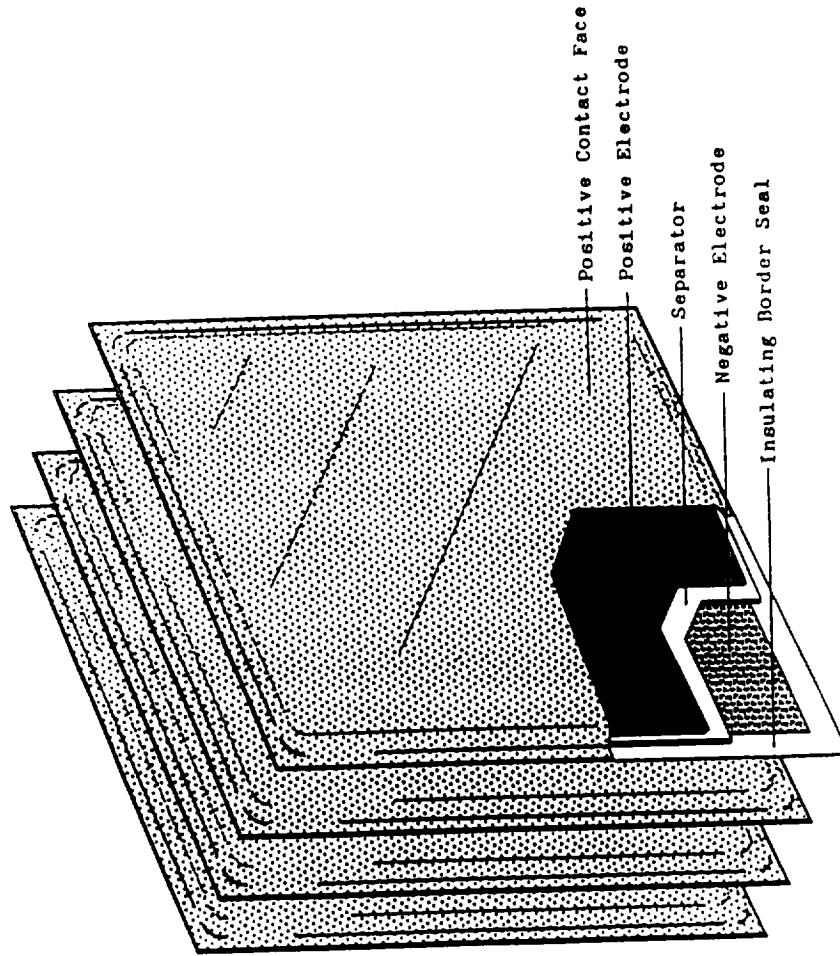
Energy Density:

<u>Present SOA</u>	<u>Program Goals</u>
77 Wh/kg	100 Wh/kg
175 Wh/l	250 Wh/l
	Challenging-Achievable

PROJECT SUMMARY

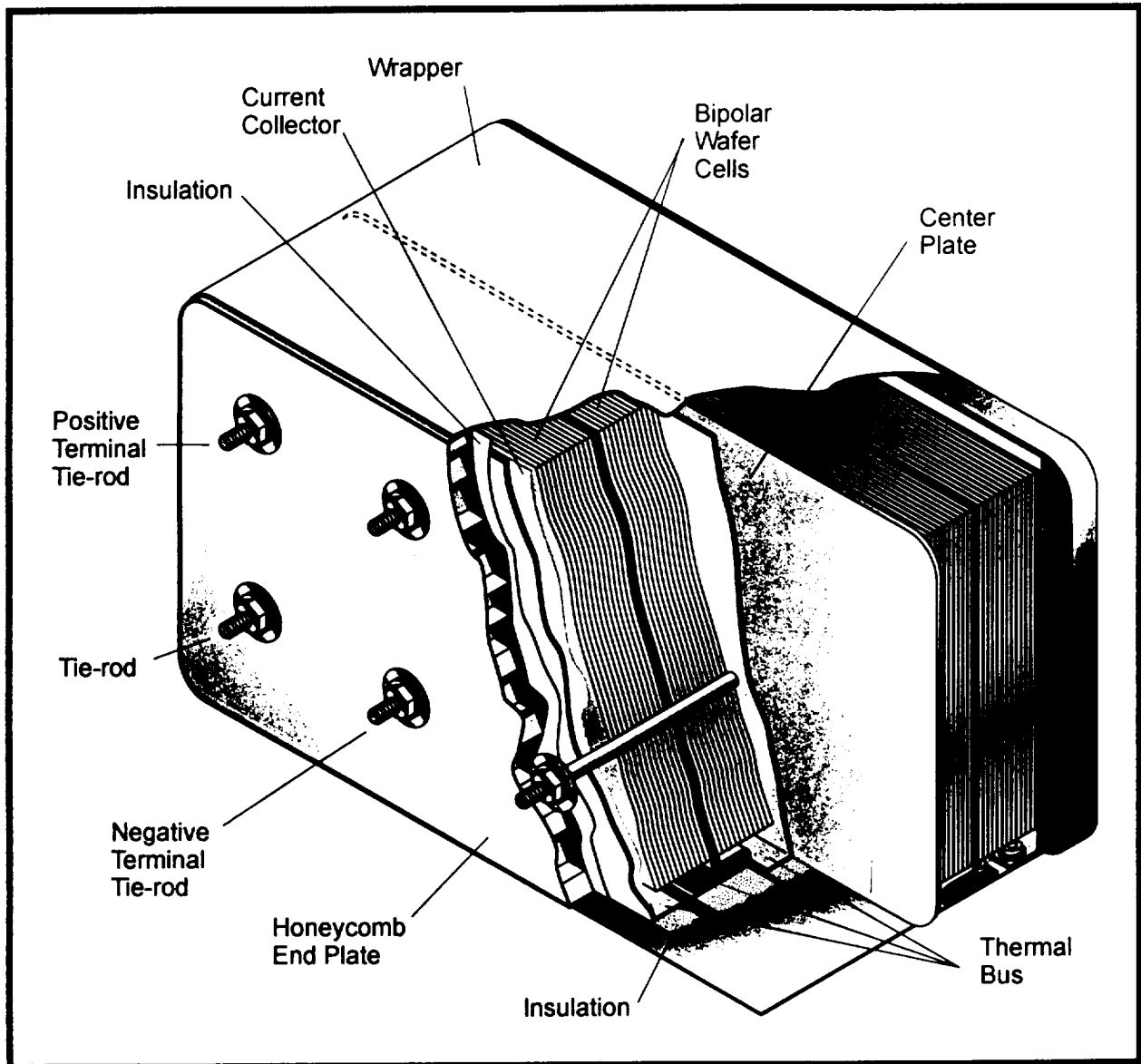
<u>Tasks:</u>	<u>Status</u>
Master Work Plan	Complete
Baseline Design	Complete
Component Development	80%
Trade Study	95%
Subscale Bipolar Batteries	66%
Preliminary Design Battery	Complete
Preliminary Boilerplate Hardware	50%
Improved Design Battery	Complete
Improved Design Prototypes	Starts 1998
Optimized Design Battery	Starts 1999
Optimized Design Prototypes	Starts 1999
Final Flight-weight Design	Starts 2000
Prototype Battery Option	Starts 2000

EEI's Stackable Wafer Cell Concept*



***U.S. Patent #5,393,617
U.S. Patent #5,552,243**

Electro Energy's Bipolar Nickel Metal Hydride Battery



Electro Energy, Inc.

Shelter Rock Lane
Danbury, CT 06810
(203) 797-2699

ADVANTAGES OF EEI WAFER BIPOLAR DESIGN

- ◆ Each cell individually sealed
- ◆ Allows repair or replacement of individual cells
- ◆ No external cell terminals
- ◆ No electrode current collectors
- ◆ Compatible with plastic bonded electrodes
- ◆ Adaptable to heat transfer fins placed in stack
- ◆ Scalable to large area, capacity, high voltage
- ◆ Automated flexible manufacturing
- ◆ Improved energy and power density
- ◆ Lower cost

COMPONENT DEVELOPMENT

Component Testing

- ◆ Three inch square bipolar wafer cell construction
- ◆ Test against baseline Ni or MH electrode with the baseline separator
- ◆ Vented, flooded cells
- ◆ Sealed starved electrolyte cells
- ◆ 50% DOD LEO cycles (55 min. Charge, 35 min. discharge)
- ◆ Charge Control:
 - Timed charge & discharge
 - 5% overcharge
- ◆ Build & test sample set cells in quadruplicate
- ◆ Build & test control cells in duplicate
- ◆ DPA Critical Experiments

COMPONENT DEVELOPMENT (cont'd)

Nickel Electrode Development

- ◆ Advanced EEI treated nickel compounds:
 - Coating Thickness - Suppliers: Tanaka, Skerrit
 - Alloying
 - NiOH Particle Size
 - Microstructure
- ◆ EEI plastic bonded nickel construction
- ◆ Sintered porous nickel
- ◆ Electrochemically impregnated fiber of foam plaque
- ◆ Pasted fiber of foam plaque

Resource Allocation

- ◆ EEI, EIP, RU

COMPONENT DEVELOPMENT (cont'd)

Metal Hydride Electrode Development

- ◆ Incremental Improvements of AB₅:
Refractor Metal Substitution Alloys
High Stoichiometric Ratio Alloy with Mo and/or Si Additions
Modified RP40 Grade Alloy
- ◆ Radical Improvements of Rechargeable Battery Alloys
Mg₂Ni Alloys
Li₂Mg₁₇ Alloys
- ◆ Plastic Bonded & Sintered AB₅ (LaNi₅) Construction
- ◆ Iron Titanium (FeTi) Doped Magnesium Compounds
- ◆ Protective Alloying or Coating

Resource Allocation

- ◆ EEL, EPI, Rh-P

COMPONENT DEVELOPMENT (cont'd)

Testing Summary

- ◆ **Up to 7000 LEO Cycles @ 50% DOD (Tripled Cycle-Life)**
- ◆ **Up to 9200 LEO Cycles @ 25% DOD (1.18 volts @ EOD)**
- ◆ **340 Sets of Cells Built and Tested - About 800 Cells**
- ◆ **Computer Controlled Automated Test Systems**
- ◆ **EEI and EPI Ni Electrodes**
- ◆ **EEI and EPI MH Electrodes**

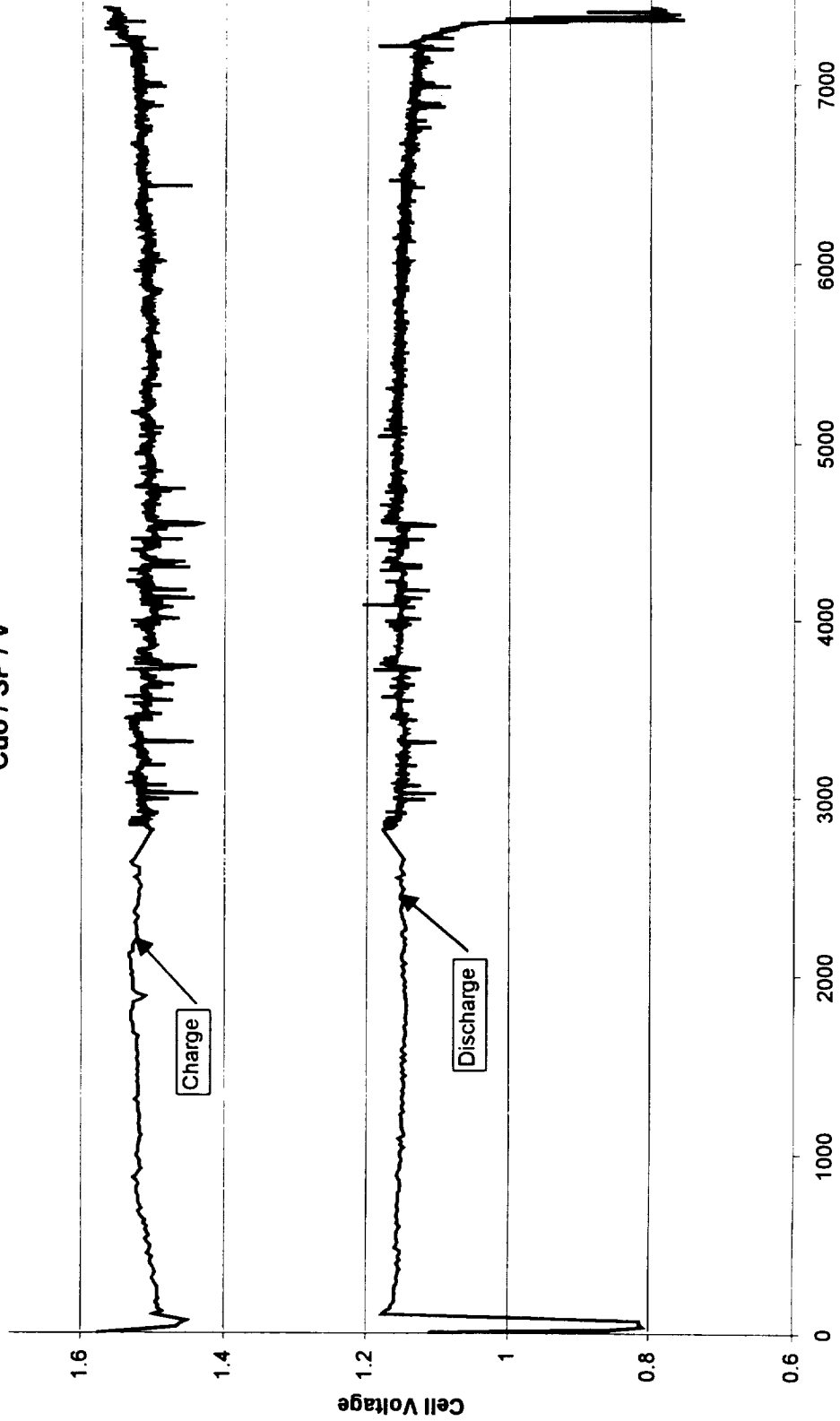
Replanned as a Four Year Task

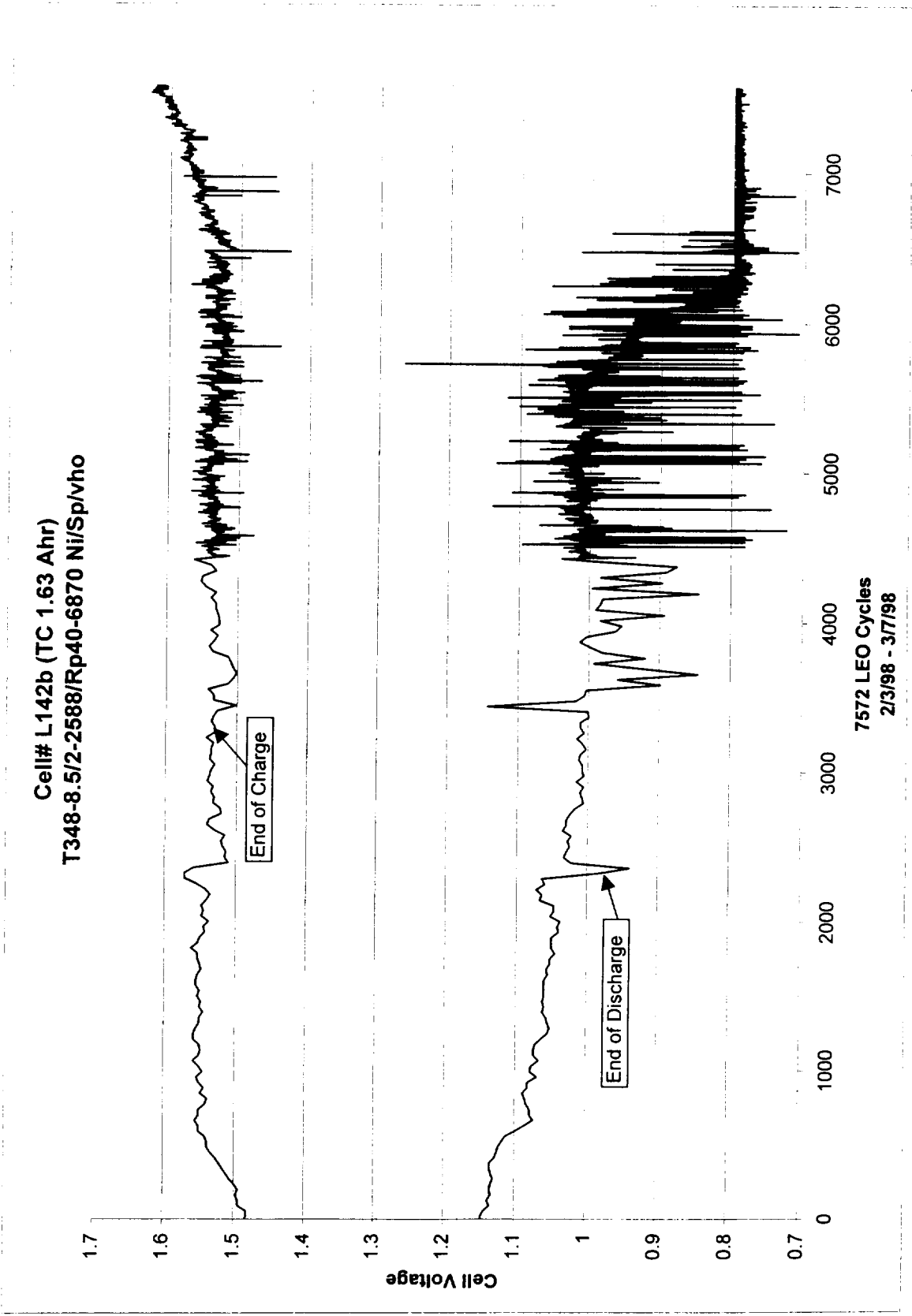
COMPONENT DEVELOPMENT (cont'd)

Foil Cell Design

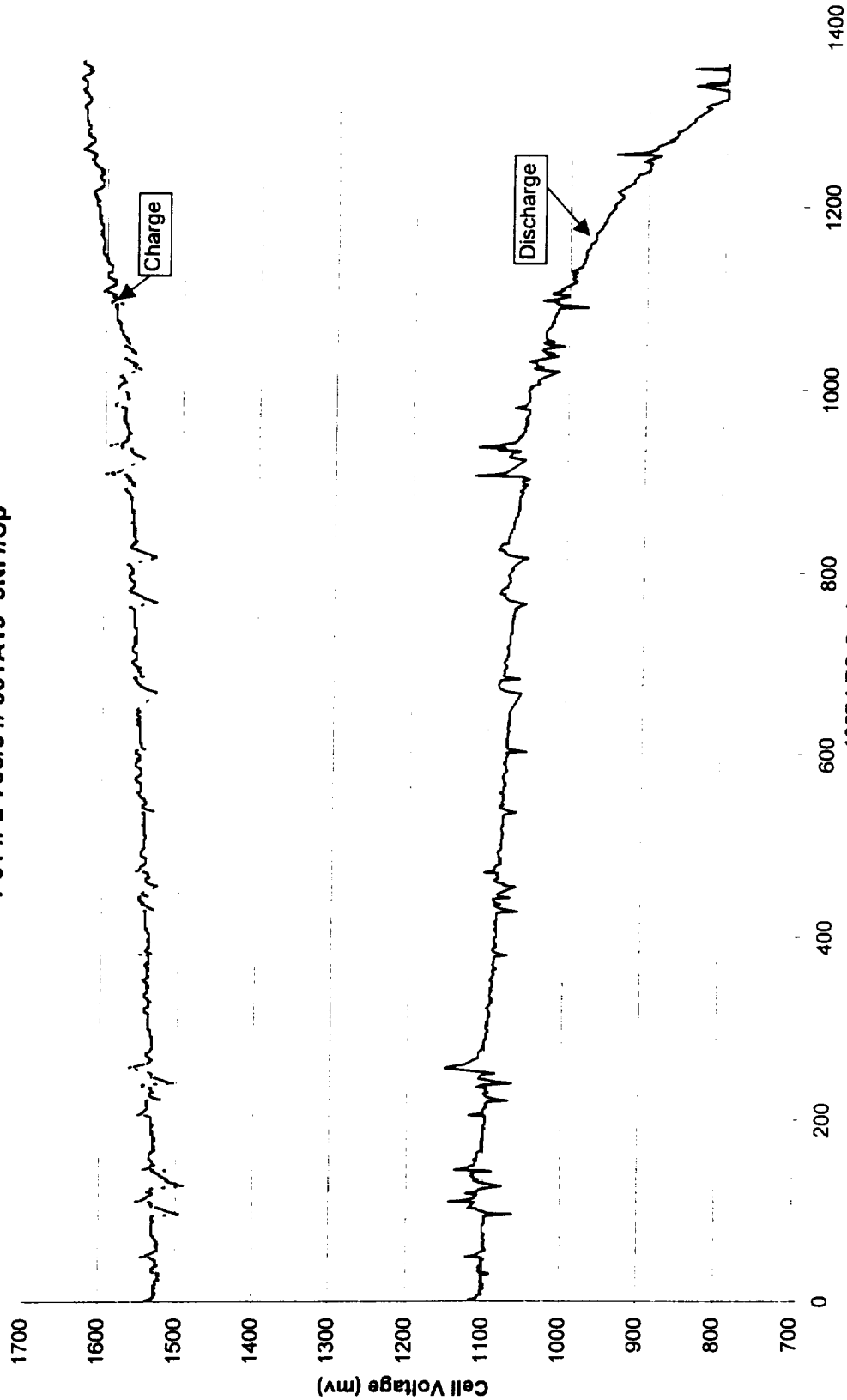
- ◆ Developed a wafer cell using nickel foil instead of carbon plastic
- ◆ Advantages:
 - Lower contact resistance between cells
 - No permeation from sealed cells
 - Stronger wafer package
 - Higher lateral thermal conductivity
 - Long Life
- ◆ Disadvantages
 - Increases Cell Assembly Labor
 - Increases Weight (slightly)
- ◆ Vented Flooded
 - L175a - 7,400 LEO cycles
 - L142b - 7,000 LEO cycles
- ◆ Sealed
 - L225b - 2,600 LEO cycles
 - L229f - 2,120 LEO cycles

Cell # L175a (Th Cap 1.0 Ahr)
P37 / 2-2588 / RP40-6870
Cuo / SP / V

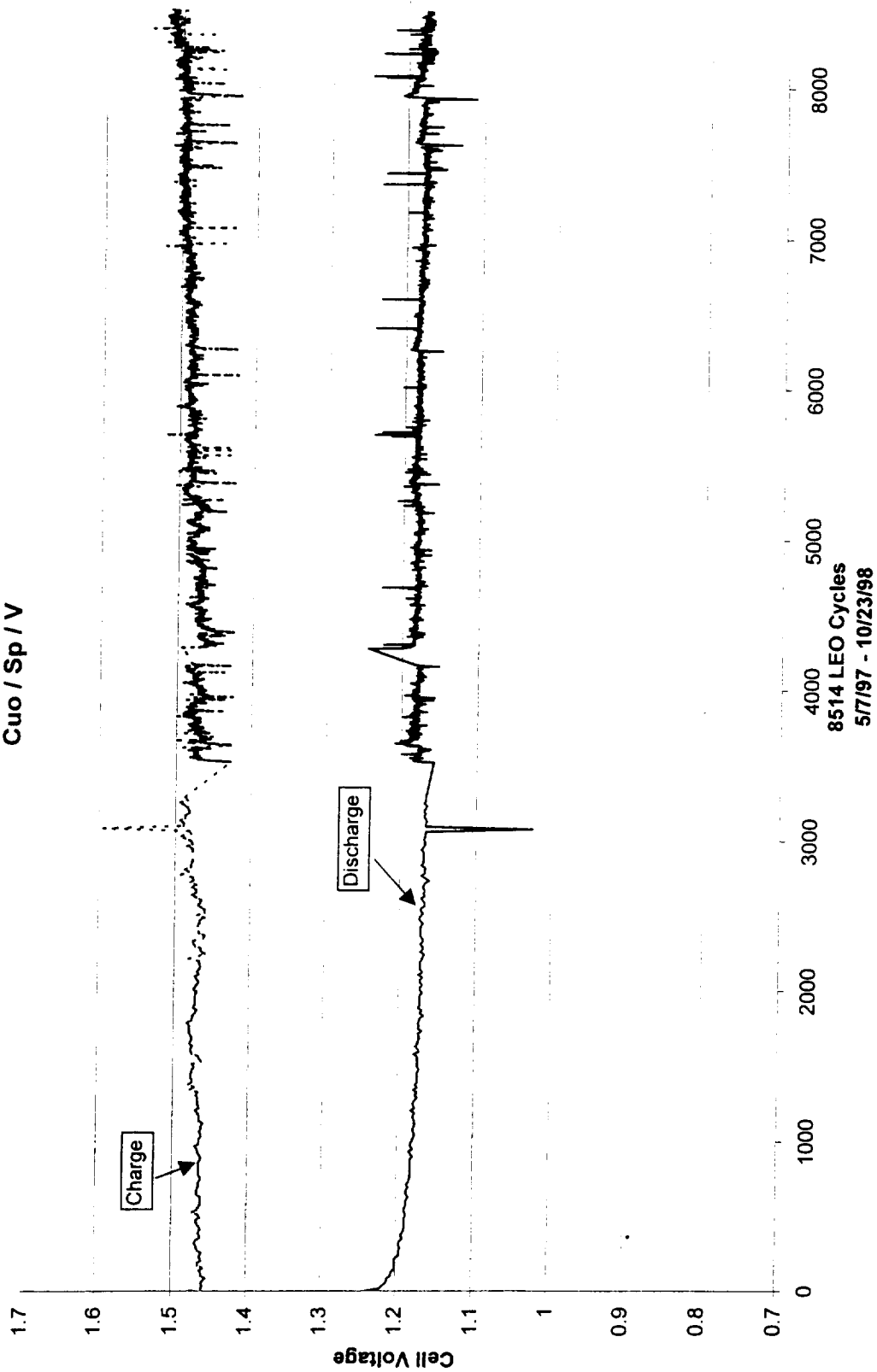




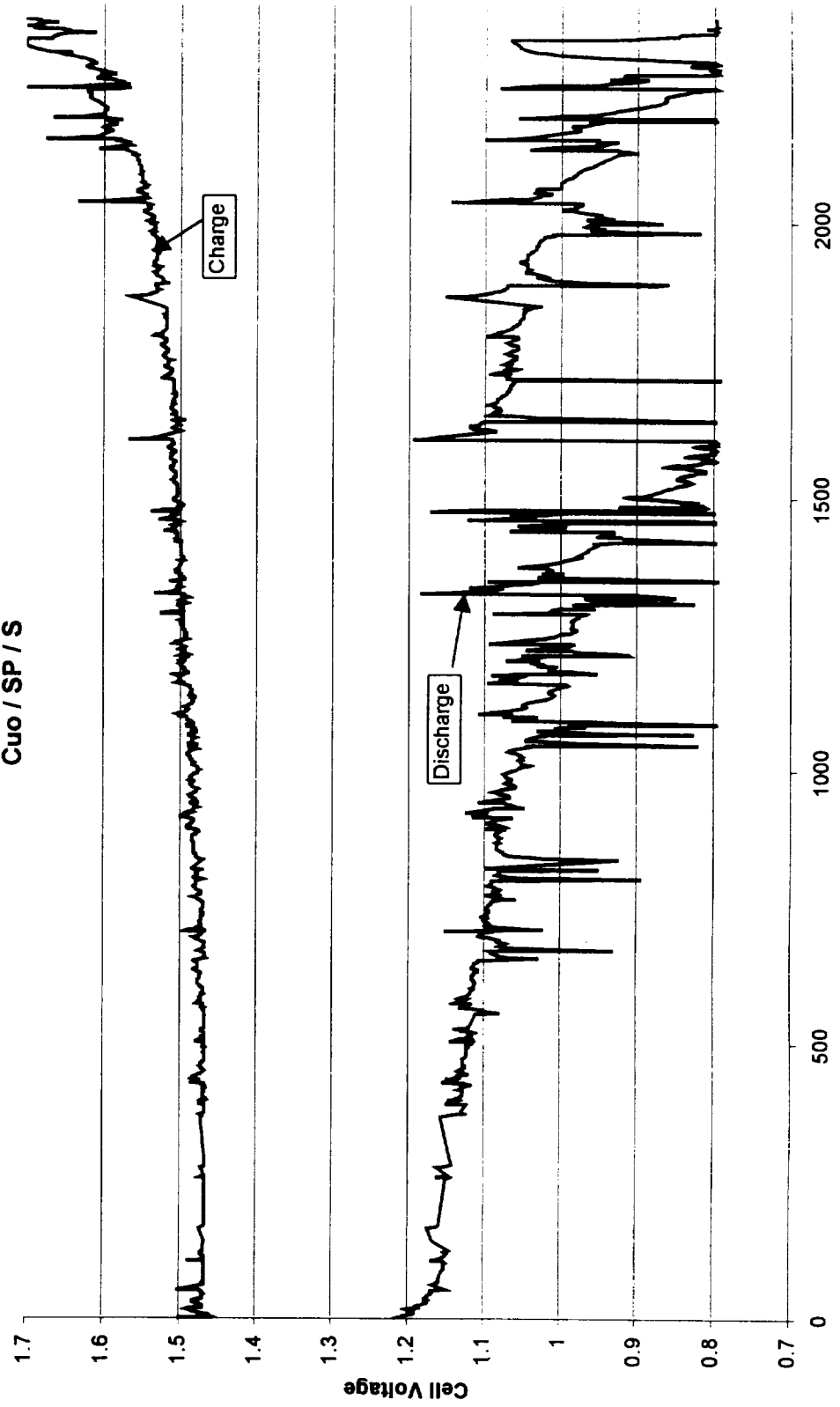
Cell # L294a (Th Cap 16.92)
P61 // 2-700/9 // 90TA19 9Ni //Sp



Cell # L173b
P37 / 2-2588 / RP40-6870
Cuo / Sp / V



Cell # L288a (Th Cap 1.62 Ahr)
P50 / 2-2588 / RP40
CuO / SP / S



2370 LEO Cycles
5/12/98 - 9/23/98

COMPONENT EVALUATION IN BIPOLAR CONFIGURATION

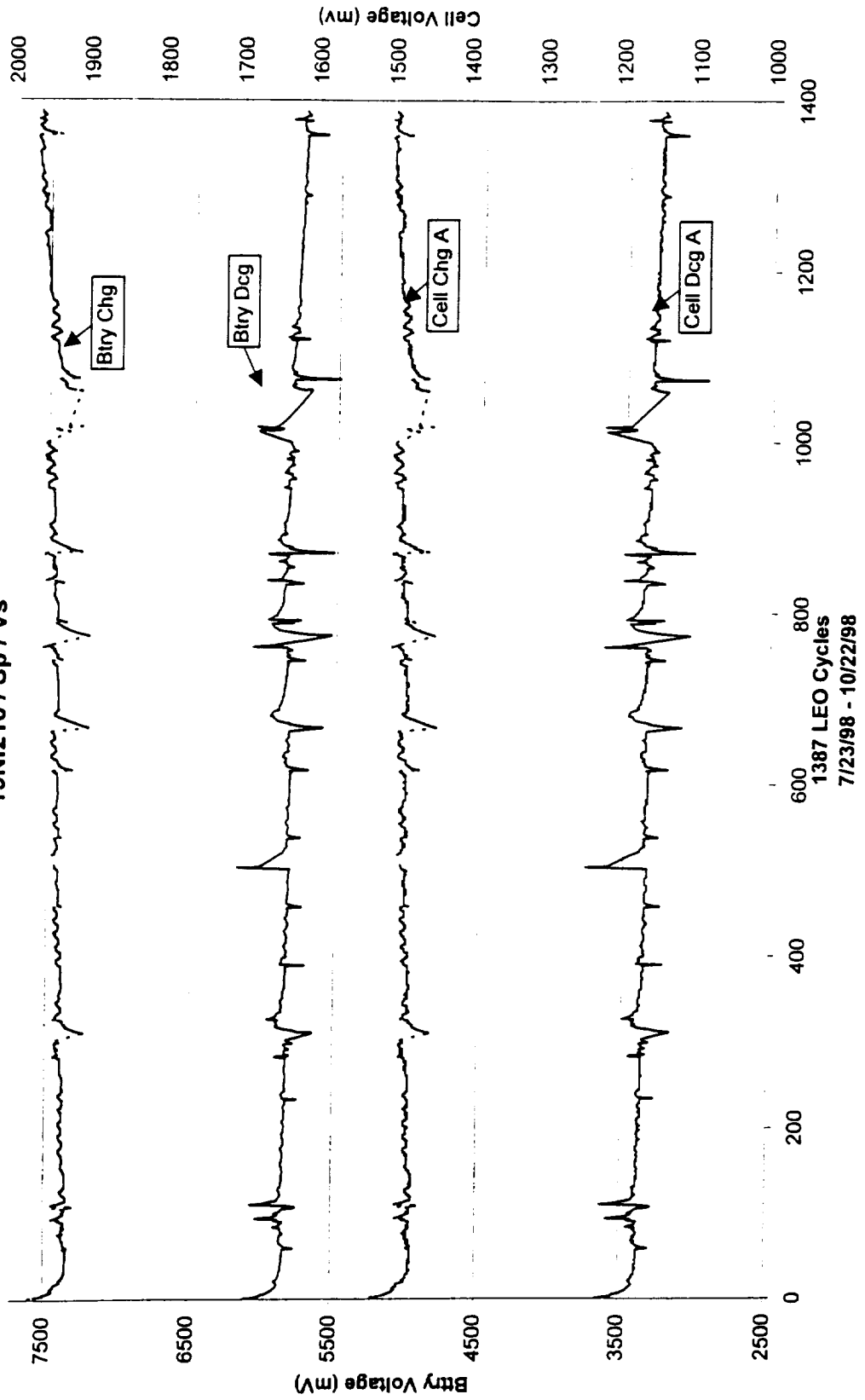
Subscale Bipolar Batteries

- ◆ Assemble 5-cell Stacks
- ◆ Construct Three Inch Square, Sealed, Starved Electrolyte Samples
- ◆ Test Three Top Ni Electrodes Against a Baseline MH
- ◆ Test Three Top MH Electrodes Against a Baseline NI
- ◆ Test for 500 to 1000 LEO Cycles at 50% DOD
- ◆ Perform DPA EEI & RU

Test Status:

- ◆ On-going: 1,100 LEO Cycles (to date) at 50% DOD; 1.15V @ EOD

Cell#319ae (TC 1.62 Ahr)
P60 // 2-700/9 // 80TA19
19Ni210 / Sp / Vs



TRADE STUDY

Rules Based Technology

- ◆ Vary all electrode and cell components
- ◆ Prismatic vs Cylindrical Configurations
- ◆ Single vs Multiple Stack

Output Summary

- ◆ Coverage - Bottom line results and 3D wireframe
- ◆ Input/Output Report - I/O by Category
- ◆ Weight Summaries - Battery, Cells and Hardware
- ◆ Layout - Front View, X-section and details
- ◆ Producibility Report
- ◆ Costed BOM
- ◆ Maintenance Guidelines

Automated Structural Analysis

- ◆ Cell Squeeze, Cyclic and Steady State Pressures
- ◆ Batch or Interactive
- ◆ 3-D wireframe model
- ◆ Area Summary
- ◆ Goodman Margins
- ◆ Contour Plot and Deformation Animation

TRADE STUDY (cont'd)

Automated Thermal Analysis

- ◆ Cold plate Location and Temperature, Power Profile
- ◆ Batch or Interactive
- ◆ 3-D wireframe model
- ◆ Area Summary
- ◆ Graphs of Max and Min Cell Temp
- ◆ Contour Plot of Cells

Trade Study Results

Battery Energy Density

- | | | |
|------------------------------------|---|---------------|
| ◆ Baseline Flight-weight Design | - | 66 Wh/kg 1995 |
| ◆ Preliminary Flight-weight Design | - | 77 Wh/kg 1997 |
| ◆ Optimized Flight-weight Design | - | 83 Wh/kg 1998 |

IMPROVED DESIGN FLIGHT-WEIGHT BATTERY SUMMARY

- ◆ **Function:**
 - The Improved Design Flight-weight Battery is a Flight-weight Battery Design for LEO Satellite applications.
- ◆ **Description:**
 - The battery is prismatic (rectangular) in shape.
 - The outer enclosure serves to contain cell stack pressure and is hermetically sealed.
 - Tie rods are used to maintain cell stack compression.
 - The battery has two (electrically parallel) 24-cell stacks (48 cells). Internal pressure is regulated by a preset relief valve.
 - There are four thermo-couples to monitor internal temperatures.

IMPROVED DESIGN FLIGHT-WEIGHT BATTERY SUMMARY (cont'd)

Design Requirements:

a. Voltage:	28 V
b. Current:	22.7 A Charge, 35.7 A Discharge
c. Power:	1 Kilowatt
d. Battery Capacity:	52 A-hr (20.8 A-hr at 40% DOD)
e. Cell Weight (wet):	157 grams
f. Cell Energy Density:	114 Wh/kg
g. Cell Capacity:	26 Amp hours
h. Current Density:	77 mA/cm ²
i. Cycle Life:	30,000
j. Operating Temperature:	-10° C to +25° C
k. Overcharge:	5%
l. Electrical Insulation Resistance:	> 1 megohm @ 100 VDC
m. Cooling:	Conductive baseplate
n. Maximum Temperature Change:	25° C above cold plate
o. Thermal Generation:	128 Watts Average

IMPROVED DESIGN FLIGHT-WEIGHT BATTERY SUMMARY (cont'd)

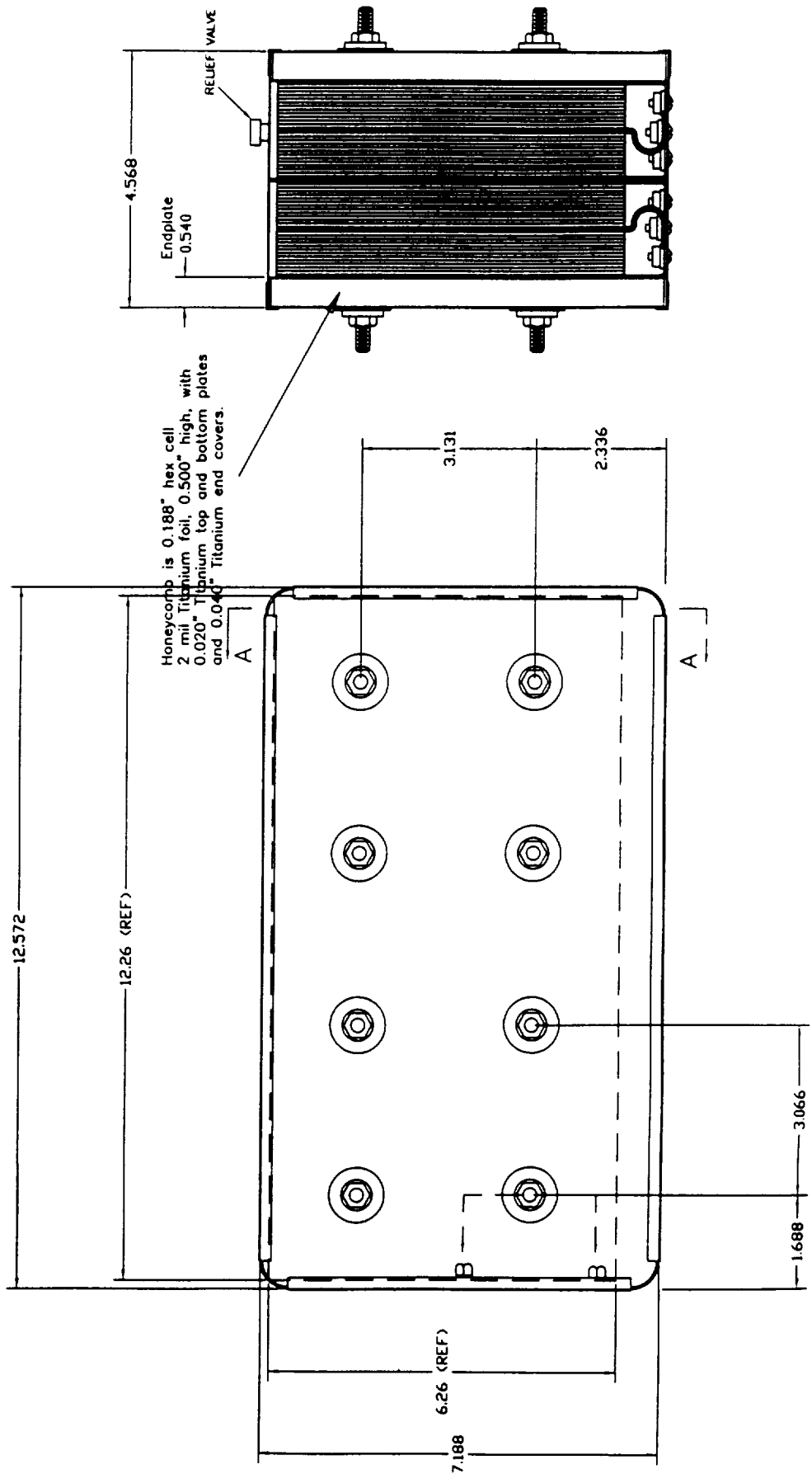
Size

- ◆ Overall dimensions of the battery case are approximately 12.6" wide by 7.2" high by 4.7" deep
- ◆ The Nickel-Metal Hydride cells (12) are approximately: 0.066" thick by 6.56" high by 12.56" wide

Materials

- ◆ Enclosure - Titanium
- ◆ Thermal Plates - Ni plated aluminum
- ◆ Cells:
 - The lightweight Nickel electrode is plastic bonded treated Nickel Hydroxide.
 - The separator consists of one layer of Pellon 2588.
 - The Metal Hydride electrode is EEI plastic bonded RP40.
 - The Electrolyte is 30% by weight Potassium Hydroxide solution with 1% Lithium Hydroxide.
 - Nickel Foil Cell Face.
 - Epoxy Insulating Frame.
- ◆ Tie Rods - Titanium
- ◆ Current Collectors - Nickel Foil

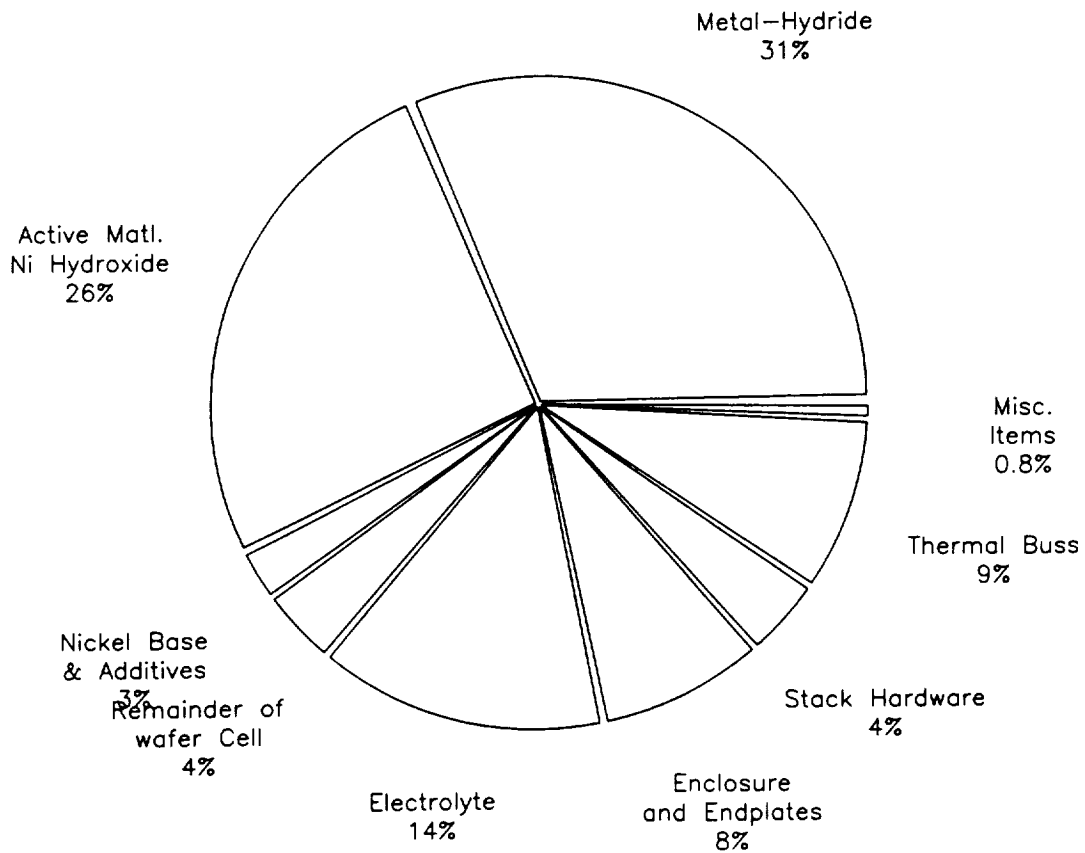
IMPROVED DESIGN FLIGHT-WEIGHT BATTERY SUMMARY (cont'd)



Battery Weight Summary

Battery Weight Table			
Item	Wt.(g)	Wt.(lbs)	Wt.%
Metal-Hydride	5500.0	12.1	31.3
M-H Electrode Additives	27.6	0.1	0.2
Nickel Base & Additives	442.6	1.0	2.5
Active Material Ni Hydroxide	4552.9	10.0	25.9
Electrolyte	2534.1	5.6	14.4
Remainder of Wafer Cell	679.9	1.5	3.9
Thermal Buss	1528.2	3.4	8.7
Stack Hardware	700.2	1.5	4.0
Enclosure and Endplates	1447.6	3.2	8.2
Miscellaneous Items	134.7	0.3	0.8
Total	17548.0	38.7	100

Battery Weight Distribution



Electro Energy, Inc.
Design Automation Associates
Version: R1.15

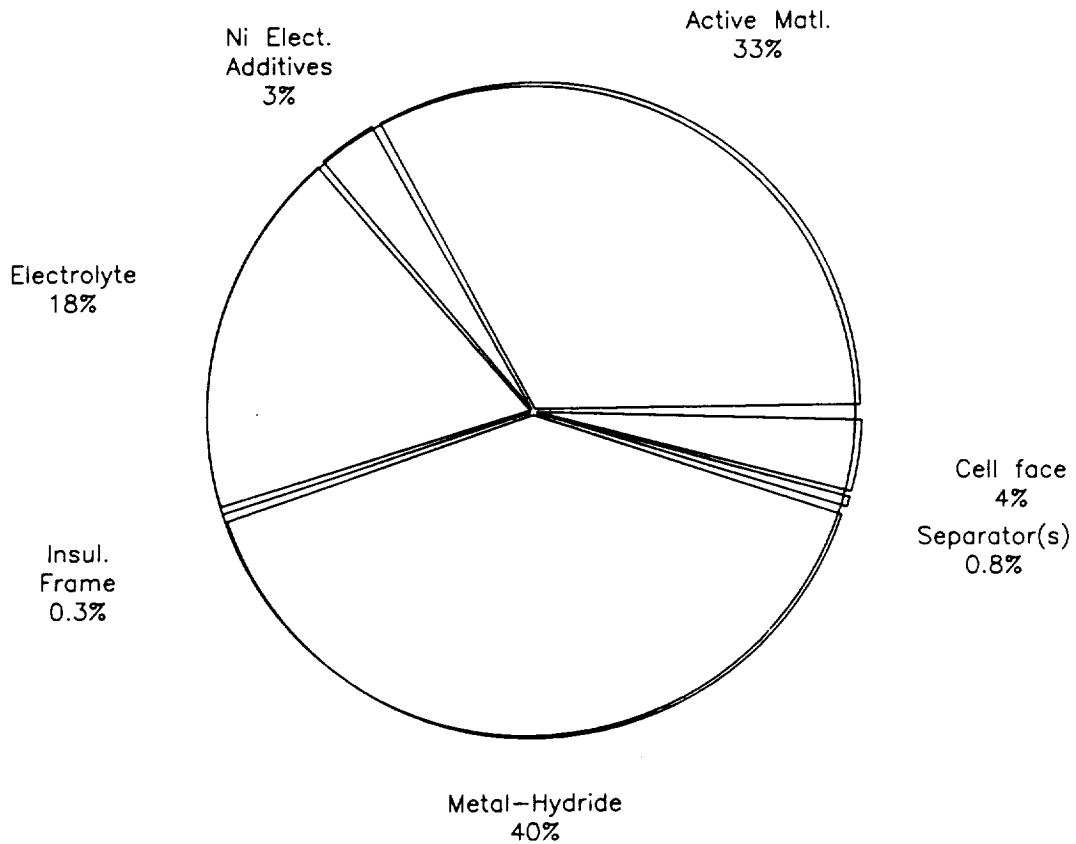
Page 6 of 13

NASA Lewis
Research Center
Date: October 20, 1998

Wafer Cell Weight Summary

Wafer Cell Weight Table		
Item	Wt.(g)	Wt.%
Active Material	94.9	33.1
Ni Electrode Additives	9.2	3.2
Metal-Hydride	114.6	40.0
M-H Electrode Additives	0.6	0.2
Washer	0.4	0.1
Separator(s)	2.2	0.8
Electrolyte	52.8	18.4
Insulating Frame	0.8	0.3
Cell Face	10.8	3.8
Total	286.2	100

Wafer Cell Weight Distribution



Electro Energy, Inc.
 Design Automation Associates
 Version: R1.15

Page 7 of 13

NASA Lewis
 Research Center
 Date: October 20, 1998

PRELIMINARY DESIGN BOILERPLATE BATTERY SUMMARY

- ◆ **Function:**
 - The Preliminary Design Boilerplate Battery is a prototype of the Flightweight Battery Design for Development Testing.

- ◆ **Description:**
 - The battery is prismatic (rectangular) in shape.
 - The outer enclosure serves to contain cell stack pressure and is sealed with o-rings and a gasket.
 - Tie rods are used to maintain cell stack compression.
 - The battery has one 12-cell stack. Internal pressure is regulated by a preset relief valve.
 - There are thermo-couples to monitor internal temperatures.

PRELIMINARY DESIGN BOILERPLATE BATTERY SUMMARY (cont'd)

Size

- ◆ Overall dimensions of the battery case are approximately 9.4" wide by 2.5" high by 13.6" long
- ◆ The Nickel-Metal Hydride cells (12) are approximately: 0.085" thick by 6.56" high by 12.56" long

Materials

- ◆ Enclosure - Ni plated aluminum
- ◆ Thermal Plates - Ni plated aluminum
- ◆ Cells:
 - The lightweight Nickel electrode is plastic bonded treated Nickel Hydroxide.
 - The separator consists of two layers of Pellon 2588.
 - The Metal Hydride electrode is EEl plastic bonded RP40.
 - The Electrolyte is 30% by weight Potassium Hydroxide solution with 1% Lithium Hydroxide.
 - Nickel Foil Cell Face.
 - Epoxy Insulating Frame.
- ◆ Tie Rods - Zinc plated steel
- ◆ Current Collectors - Nickel Foil
- ◆ Seals - Neoprene

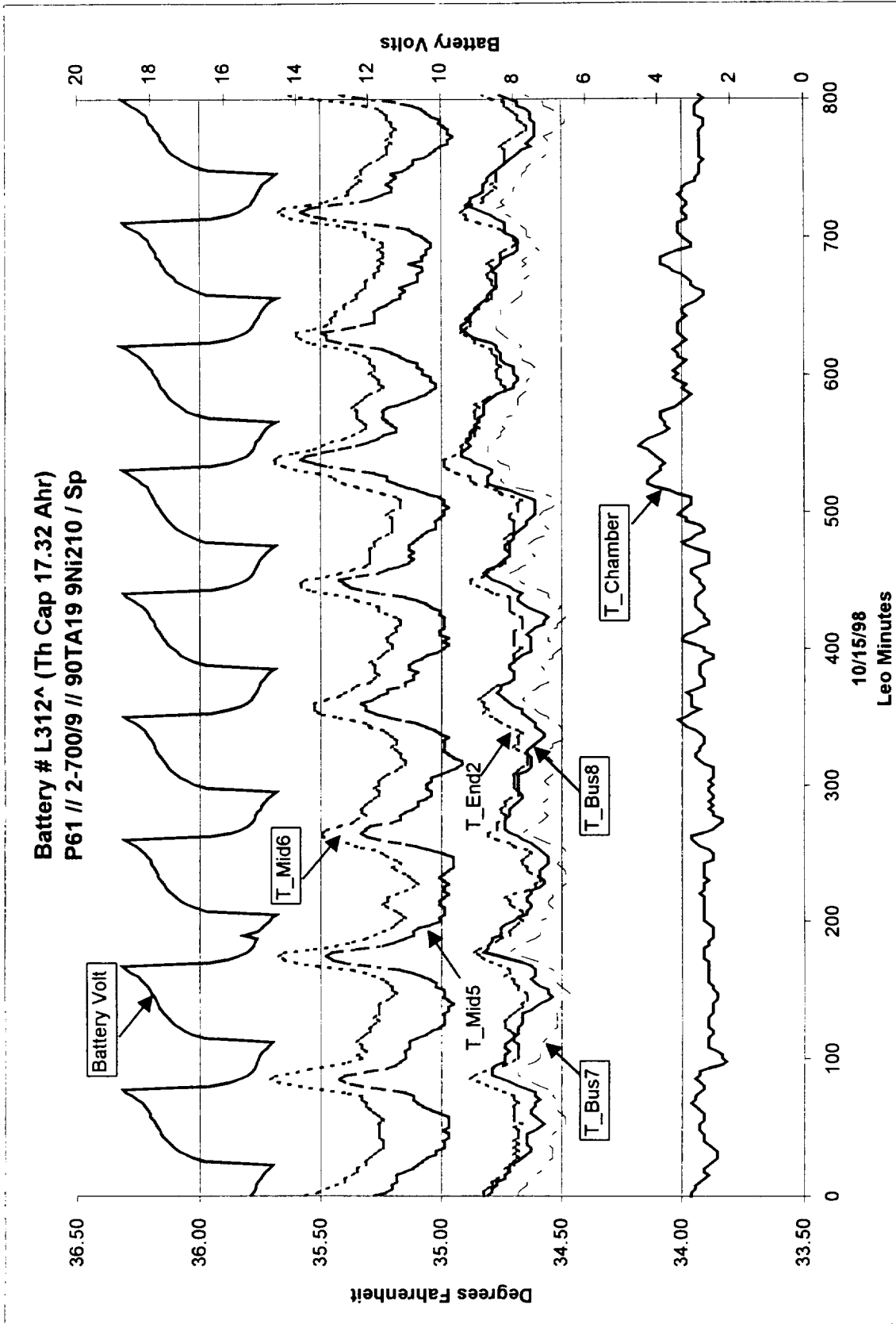
PRELIMINARY DESIGN BOILERPLATE BATTERY SUMMARY (cont'd)Design Requirements:

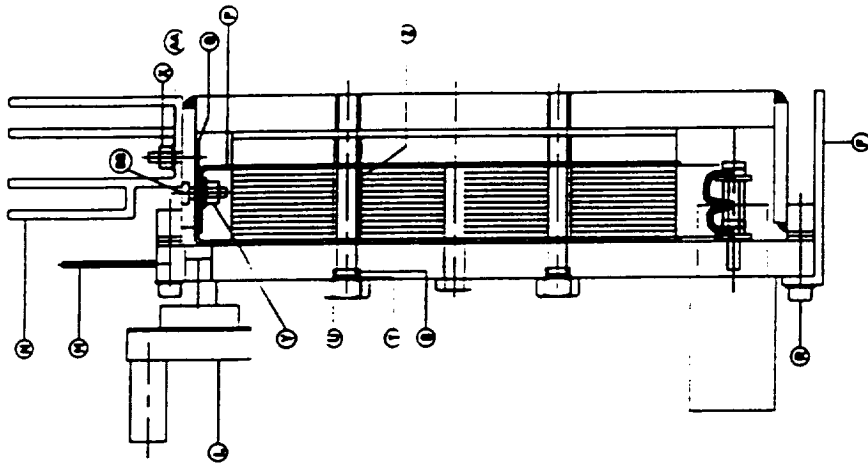
a.	Voltage:	14.4 V
b.	Current:	7.9 A Charge, 11.9 A Discharge
c.	Power:	.249 Kilowatt
d.	Battery Capacity:	17.3 A/hr (6.9 A/hr at 40% DOD)
e.	Cell Weight (wet):	313 grams
f.	Cell Energy Density:	63 Wh/kg
g.	Cell Capacity:	17.3 Amp hours
h.	Current Density:	26 mA/cm ²
i.	Cycle Life:	TBD
j.	Operating Temperature:	-10° C to +25° C
k.	Charge Retention:	TBD
l.	Overcharge:	5%
m.	Electrical Insulation Resistance:	> 1 megohm @ 100 VDC
n.	Cooling:	Conductive baseplate
o.	Maximum Temperature Change:	25° C above cold plate
p.	Thermal Generation:	32 Watts Average

PRELIMINARY DESIGN BOILERPLATE BATTERY TASK

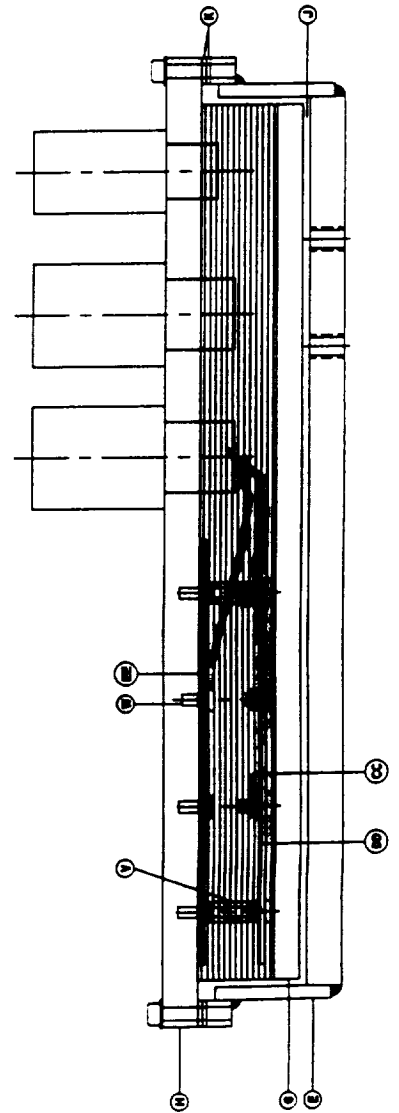
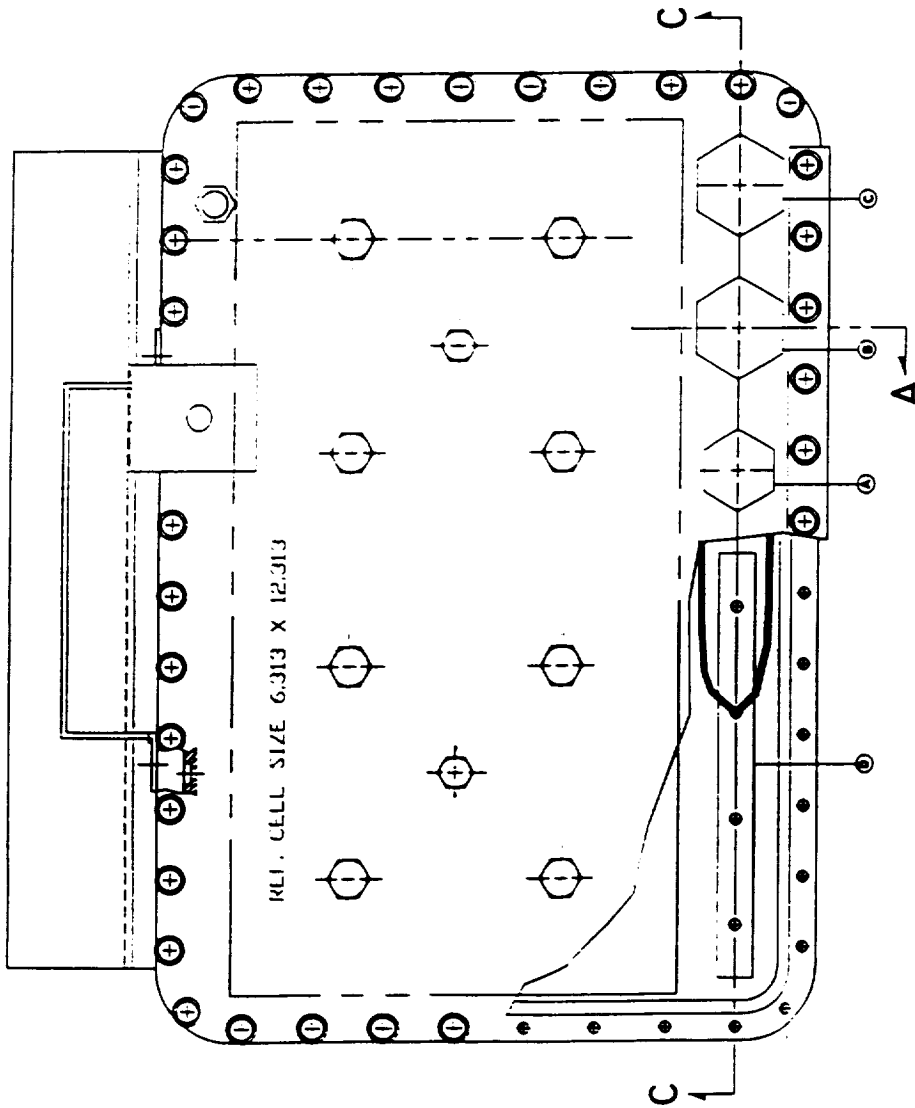
- ◆ Test Results:
 - Theoretical Capacity: 17.3 A-hrs
 - Formation Capacity: 13.0 A-hrs
 - LEO Test (40% DOD, 5% OC): On-going

- ◆ Future Work:
 - Build a second PDBB (Nov. 98)
 - DPA first PDBB (1999)





SECTION A-A



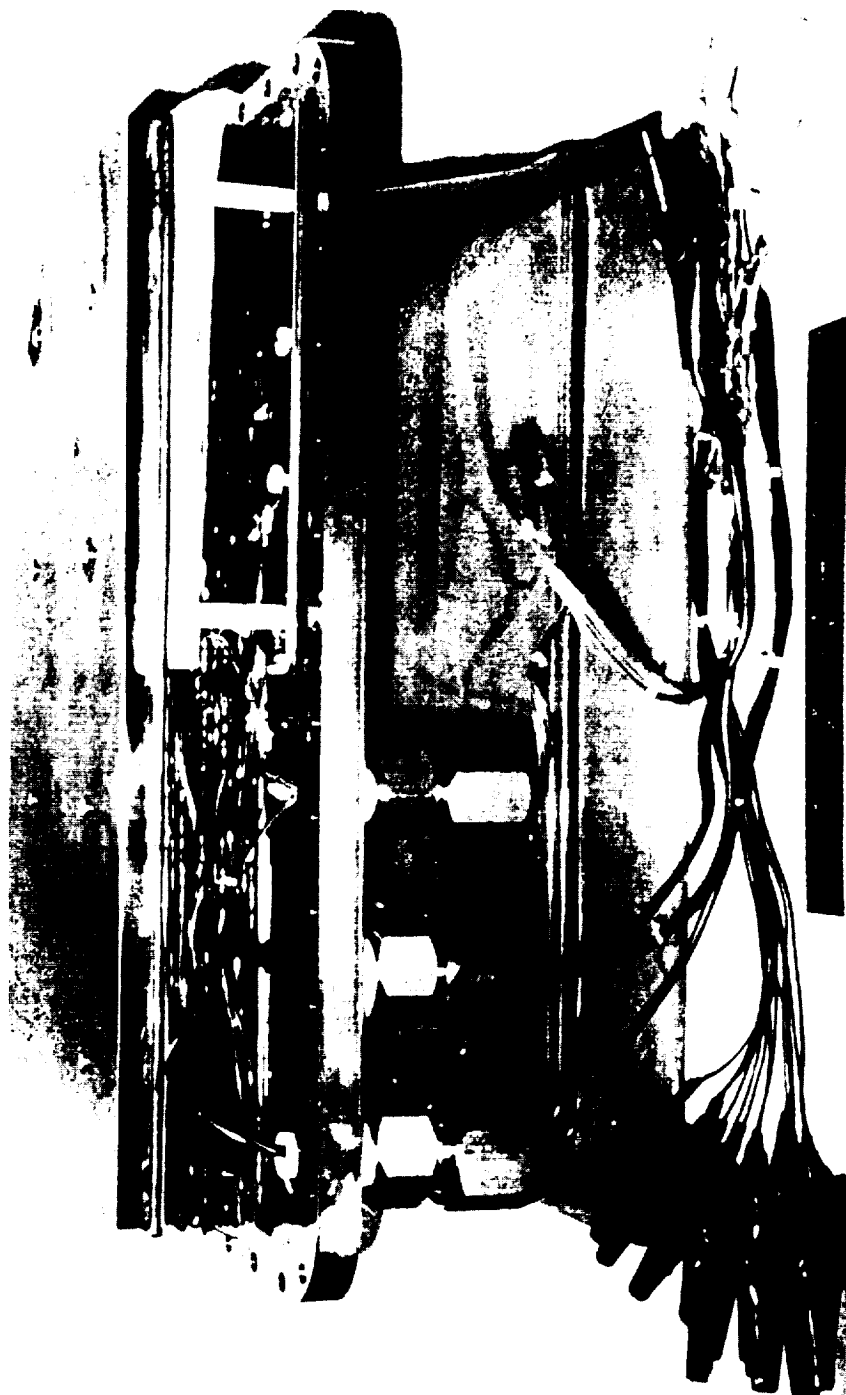
SECTION C-C

PART LIST

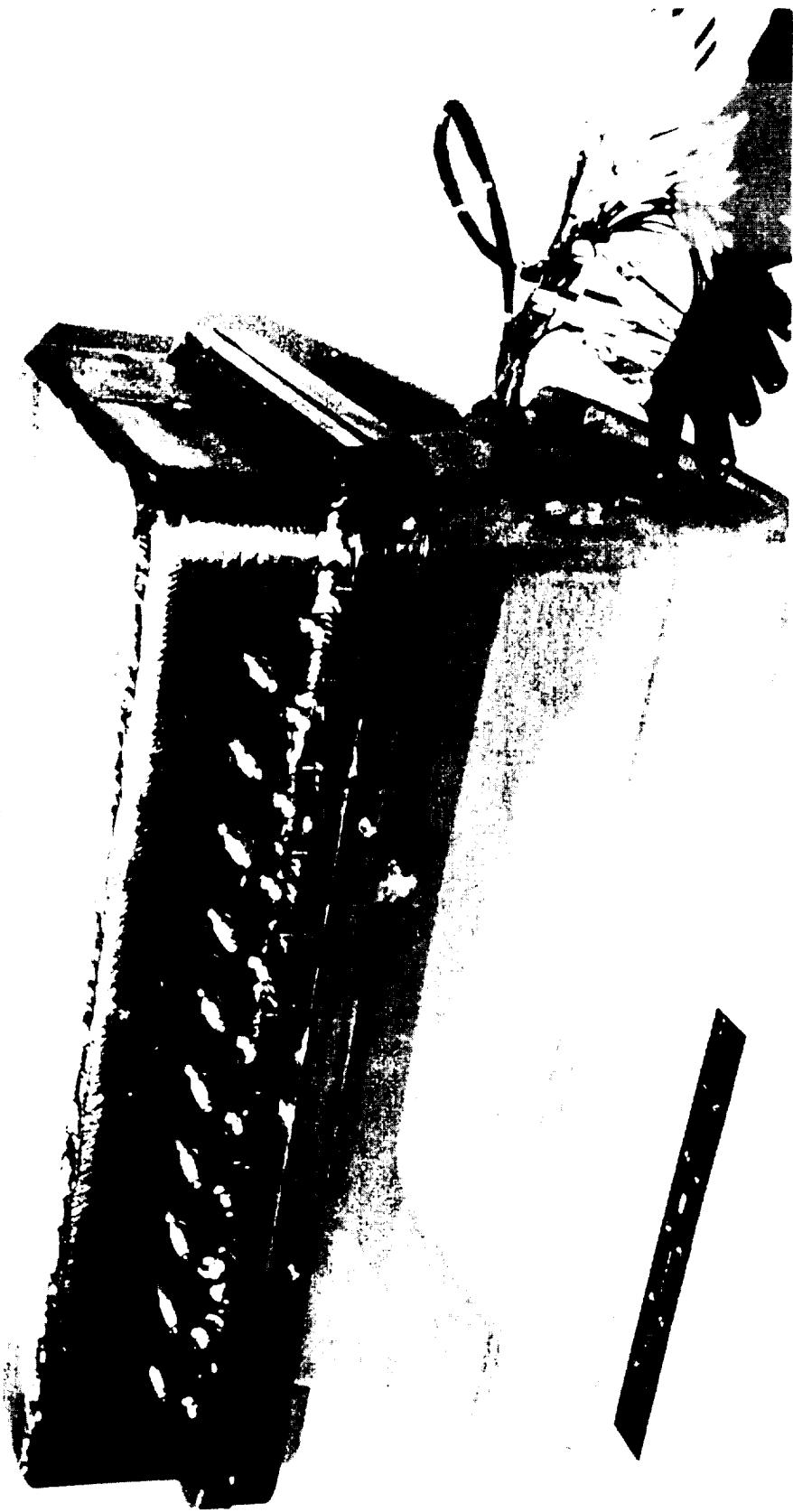
ITEM	DESCRIPTION	DWG. OR PART NUMBER	QTY
A	1" SEALED ELECTRICAL FEED	NBC-A-040-8	1
B	3/4" SEALED ELECTRICAL FEED	PL-12-2	1
C	3/4" SEALED ELECTRICAL FEED	NBC-S-032-16	1
D	BUS BAR	DA-018	1
E	HOUSING WELDMENT	DA-004	1
F	STAND	DA-016	1
G	FLOATING PLAT	DA-011	1
H	COVER	DA-005	1
J	FOAM RUBBER FILL GAP		1
K	GASKETS	DA-019	2
L	OMEGA INDUSTRIAL PRESSURE TRANSDUCER		1
M	HANDLE	MCMMASTER CARR #169A11	1
N	COOLING FAN	DA-015	1
P	THERMAL BUS	DA-013	1
Q	THERMAL BUS, STUD DESIGN	DA-014	1
R	#10-32 SOCKET HEAD SCREW		44
S	O'RING		8
T	BACK UP O'RING		8
U	#1/4-20 BOLT		8
V	#8-32 NYLON STANDOOF 1/4" HEX X 1"		2
W	#8-32 X 3/75 SCREW		6
X	#8 NUT		11
Y	#6 SELF LOCKING NUT		11
Z	#1/4-20 TIE RODS		8
AA	#8 WASHER		10
BB	SELF SEALING SCREWS		11
CC	#8 NUT		2
DD	#8 WASHER		2
EE	12 GA WIRE LUG		2

1411

Preliminary Design Bipolar Ni-MH/Lead-Acid Battery



Primary Image of the Battery

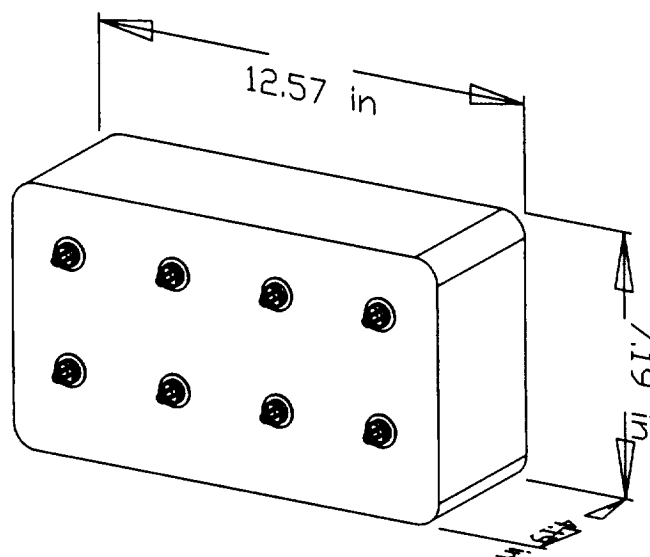


100 W-HR/kg APPROACHES

- (1) Develop a MH Alloy that has 500 mA-hr/g and 5-year LEO Life
 - ◆ On-going
- (2) Develop a Ni Electrode that gives 1.25 electron transfers and a 365 mA-hr/g and 5-year LEO Life

Bipolar Nickel Metal Hydride Battery Design System

User: MnStk+500 mAh/g MH
Power: 1.0 kW
Battery Capacity: 52.1 Ah



	<u>Metric</u>	<u>SAE</u>
Specific Energy:	99.3 Wh/kg	218.9 Wh/lb
Energy Density:	234.9 Wh/l	8.3 Wh/ft ³
Overall Weight:	14.7 kgs	32.4 lbs
Cell Weight:	11.0 kgs	24.3 lbs

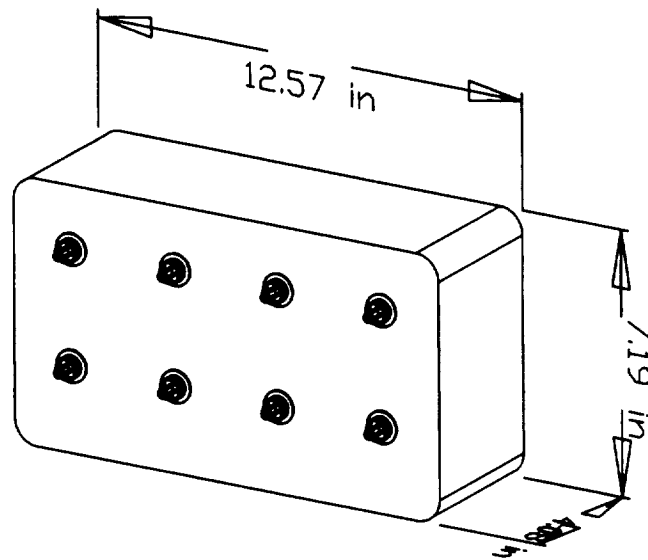
Electro Energy, Inc.
Design Automation Associates
Version: R1.15

Page 1 of 13

NASA Lewis
Research Center
Date: October 20, 1998

Bipolar Nickel Metal Hydride Battery Design System

User: Min Stk+125% Theo Ni+365MH
Power: 1.0 kW
Battery Capacity: 52.1 Ah



	<u>Metric</u>	<u>SAE</u>
Specific Energy:	100.2 Wh/kg	220.9 Wh/lb
Energy Density:	241.5 Wh/l	8.5 Wh/ft ³
Overall Weight:	14.6 kgs	32.1 lbs
Cell Weight:	10.7 kgs	23.7 lbs

Electro Energy, Inc.
Design Automation Associates
Version: R1.15

Page 1 of 13

NASA Lewis
Research Center
Date: October 20, 1998

Status of EEI's Bipolar Ni-MH Battery Project

The Bipolar Ni-MH Battery Project has:

- ◆ **Demonstrated the Wafer cell concept in multi-cell stacks on the LEO test regime**
- ◆ Demonstrated the advantages of the bipolar wafer cell concept in multi-cell stacks
 - Allows repair or replacement of individual cells
 - No external cell terminals
 - No electrode current collectors
 - Adaptable to heat transfer fins placed in stack
 - Scalable to large area and high capacity
 - Improved energy and power density
 - Lower cost plastic bonded electrodes
- ◆ **Tripled the LEO cycle-life of vented-flooded bipolar wafer cells**
 - Vented-flooded cells: 7400 LEO cycles @ 50% DOD
 - 9200 LEO cycles @ 25% DOD (EODV=1.18 V)
- ◆ **Doubled the LEO cycle-life of sealed-starved bipolar wafer cells**
 - Sealed-starved cells: 2400 LEO cycles @ 50% DOD (EODV=1.0 V)
- ◆ Completed a very high energy density bipolar battery design
 - 83 Wh/kg (37.8 Wh/lb)
 - 211 Wh/liter (7.4 Wh/ft³)

Status of EEI's Bipolar Ni-MH Battery Project (cont'd)

The Bipolar Ni-MH Battery Project will:

- ◆ Build two .5 kW, 24 cell prototype batteries with lightweight housings and 105 Wh/kg cells in 1999
 - 70 Wh/kg (31.6 Wh/lb)
 - 154 Wh/liter (5.5 Wh/ft³)

- ◆ Build two 1.0 kW, 48 cell prototype batteries with lightweight housings and 114 Wh/kg cells in 2000
 - 82 Wh/kg (37.3 Wh/lb)
 - 204 Wh/liter (7.2 Wh/ft³)

- ◆ Complete a high energy density Flight-weight Bipolar Battery Design and drawing package
 - 100 Wh/kg (45.5 Wh/lb)
 - 235 Wh/liter (8.3 Wh/ft³)

Conclusions:

- ◆ Improved active Ni and/or MH materials a required to increase the battery energy density from 83 Wh/kg to 100 Wh/kg.

Acknowledgments

Thanks and recognition needs to be given to:

- ◆ **The NASA Lewis Research Center** for having the vision, and funding the project.
- ◆ **Michelle Manzo** and **Tom Miller** for their support, technical expertise, leadership and encouragement.
- ◆ **Rhodia** (formerly Rhone-Poulenc), **Rutgers University**, and **Design Automation Associates** for their significant cost sharing contributions.
- ◆ **Dr. Bao-Min Ma** (Rh), **Charles Grun** (RU), and **Sean Sullivan** (DAA) for their significant technical contributions and excellent customer support.
- ◆ **Eagle-Picher Industries** for their technical guidance and sharing of their extensive design-to-orbit experience.
- ◆ **James DeGruson** (EPI) for his project guidance and persistent grappling for EPI resources.

