

Passive Thermal Control Challenges for Future Exploration Missions

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Background

•The President's recently announced vision refocuses attention to exploration;

•Emphasis shifts *from* Earth orbit operations *to* travel to the Moon, establishing a lunar base, and an eventual journey to Mars;

•Humans went to the Moon during Apollo, but only for short stays – a different set of challenges arises when we consider going to the Moon to stay for extended periods;

•Humans have yet to visit Mars but work on advanced programs has identified some key challenges associated with sending humans to Mars.

NOTE: This presentation will focus only on passive thermal control – thermal protection (entry-related technologies) will be covered in a separate presentation.



General

•Improved Thermo-optical Coatings:

•Low solar absorptance to infrared emittance ratio (α/ϵ) – potential use on radiators;

•Inexpensive – some current low α/ϵ coatings are very expensive;

• Easy to apply;

•Resistance to property changes (due to ultra-violet radiation);

•Resistance to atomic oxygen;

Long lifetime with stable properties;

•Can be easily maintained (in lunar- or martian-dust environment);

•Variable/user specified optical properties.

•Thermal Instrumentation:

•Inexpensive;

•Reliable;

•Robust.



General (Continued)

•Cryogenic Boil-Off:

•Improved storage of cryogens for prolonged periods.

•Insulation/Isolation Technologies:

•Vacuum panels;
•Use of aerogels;
•Multi-Layer Insulations (MLI);
•Thermal Compartments;
•Thermal Switches.

•Lunar Day/Night Survival:

- •Lunar days and nights each last ~two weeks;
- •Extreme temperatures during lunar day (> +250 deg F);
- •Lunar soil reaching high temperatures in prolonged sunlight;
- •Lunar nighttime produces surface temperatures < -300 deg F



Electronics

•With further miniaturization of electronics components, power density and the associated challenges of electronics heat dissipation will provide new challenges -- Potential needs include power reductions that keep pace with electronics miniaturization;

•Improved means of heat transfer from electronics components: •Gap Fillers; •Gaskets; •Improved interface conductance.

Involvement by the thermal community is critical early in the development process.



Analysis

•Improved modeling of systems;

•Potential for large model sizes

•Improved modeling of electronics components;

•Improved thermal environment characterization;

•Planet/moon surface environments, atmospheric extinction, diffuse sky heating components;

•Improved convective heat transfer calculations for Mars surface;

•Improved compatibility with concurrent engineering tools:

- •Thermal → Structural;
- •CFD → Thermal;
- •Orbit → Thermal;

•Etc.

•Thermal Analysis of Inflatable Structures



Environment Characterization

•Lunar and Martian Orbit Environments:

•Lunar albedo and infrared emission;

•Martian albedo and infrared emission.

•Mars Surface Environments:

•Atmosphere optical depth and dust storm characteristics;

•Diffuse sky solar and infrared heating components;

•Detailed wind profiles;

•Atmospheric temperature profiles near the surface.



Testing

•Facilities to support large-scale thermal-vacuum testing for lunar- and martian surface environments are needed:

Solar simulation;

•CO2 environment at low pressure (for martian surface simulation); •Lunar and Mars surface simulation.



Wrap-up

•Expect additional technical challenges to arise as an architecture for exploration matures;

•Key aspects of passive thermal control arise as a consequence of spacecraft integration:

• Utilizing waste heat from one system to accommodate the needs of another system;

•The entire system must function successfully as a unit.