



# Extensibility of Human Asteroid Mission to Mars and Other Destinations

P. Lopez, NASA/JSC (EA4)  
M. McDonald, NASA/JSC (EA4)

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# List of Authors

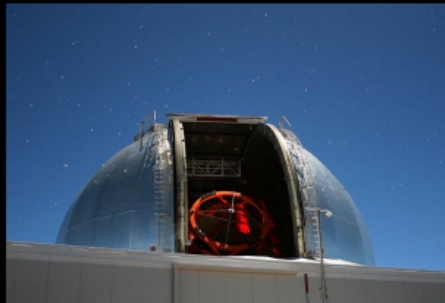


Name	Title
Mark A. McDonald	Lead, Advanced Mission Development Group, Technical Integration Office, Engineering Directorate, NASA JSC/EA4
Jose M. Caram	Deputy Manager, Technical Integration Office, Engineering Directorate, NASA JSC/EA4
Pedro Lopez	Aerospace Engineer, Advanced Mission Development Group, Technical Integration Office, Engineering Directorate, NASA JSC/EA4
Heather D. Hinkel	Rendezvous Sensor Lead, Aeroscience and Flight Mechanics Division, Engineering Directorate, NASA JSC/EG6
Jonathan T. Bowie	AEMU SE&I Lead Engineer, Space Suit and Crew Survival Systems Branch, Engineering Directorate, NASA JSC/EC5
Paul A. Abell	Lead Scientist for Planetary Small Bodies, Astromaterials Research and Exploration Science Directorate, NASA JSC/KR
Bret G. Drake	Deputy Manager, Exploration Missions and Systems Office, NASA JSC/YX
Roland M. Martinez	Aerospace Engineer, Exploration Mission Systems Office, Human Exploration Development Support, NASA JSC/YX111
Paul W. Chodas	Senior Engineer, NASA NEO Program Office, 301-121, Jet Propulsion Laboratory, California Institute of Technology.
Kurt Hack	Aerospace Engineer, Mission Design and Analysis Branch, NASA GRC, Cleveland, OH 44135, Senior Member AIAA.
Daniel D. Mazanek	Senior Space Systems Engineer, Systems Analysis and Concepts Directorate, Space Mission Analysis Branch (E402)/Mail Stop 462, NASA LaRC, Senior Member AIAA.

- Asteroid Redirect Mission Overview
- Observation Campaign Extensibility
- Asteroid Redirect Robotic Mission Extensibility
  - Solar Electric Propulsion
  - Asteroid Resource Utilization
  - Robotic Boulder Capture
  - Planetary Defense
- Asteroid Redirect Crewed Mission Extensibility
  - Automated Rendezvous Sensors
  - EVA
  - Sample Collection
  - Automated Docking
- Extensibility to Missions Post-ARCM
  - Extended Missions to Asteroid in DRO
  - Mars Phobos Mission
  - Lunar Mission
  - Capability Driven Framework
- Conclusion

## Overall Mission Consists of Three Main Segments

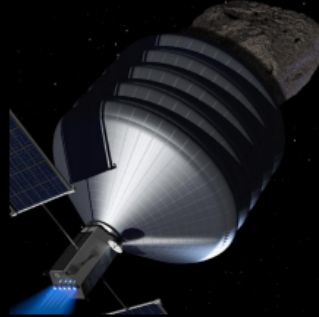
### Identify



#### **Asteroid Identification Segment:**

Ground and space based NEA target detection, characterization and selection

### Redirect



#### **Asteroid Redirection Segment:**

Solar electric propulsion (SEP) based robotic asteroid redirect to trans-lunar space

### Explore



#### **Asteroid Crewed Exploration Segment:**

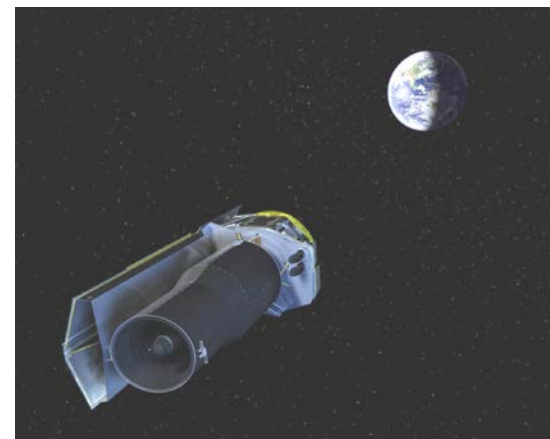
Orion and SLS based crewed rendezvous and sampling mission to the relocated asteroid

# Observation Campaign Extensibility

- Leverages off of NASA's Near-Earth Asteroid (NEA) discovery and characterization effort
- Utilizes existing ground and space-based observation assets
- New telescopes are coming online, and existing surveys are being upgraded with greater time allocations and improved cameras in support of ARM observation campaign
  - Will increase discovery rate of candidate asteroids for ARM
- These enhancements are also applicable to discovery of hazardous asteroids in general



Catalina Sky Survey Telescope



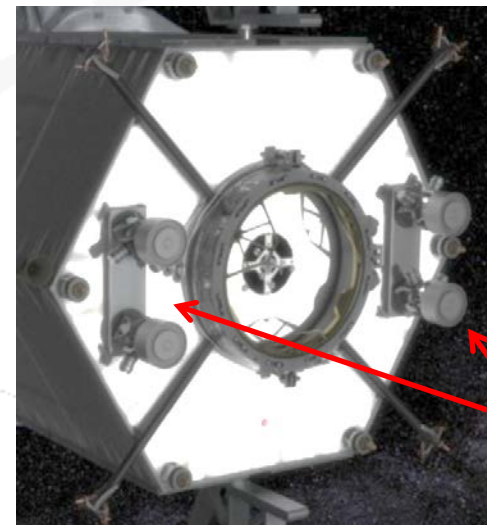
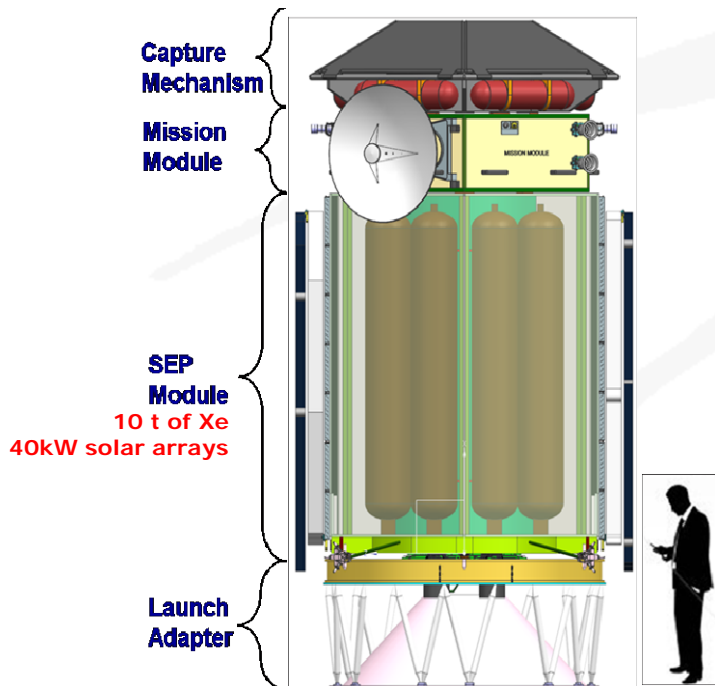
Spitzer Infrared Space Telescope



# Asteroid Redirect Robotic Mission Extensibility

## Solar Electric Propulsion (SEP)

- Capability of SEP system on ARRV could be used in other human exploration missions
  - Longer duration near-Earth missions in cislunar space
  - Cargo delivery for human Mars missions
    - Will require higher power for SEP systems; addressed via solar array systems being demonstrated under current STMD effort (extensible to 125 kW per wing via modularity and scalability)

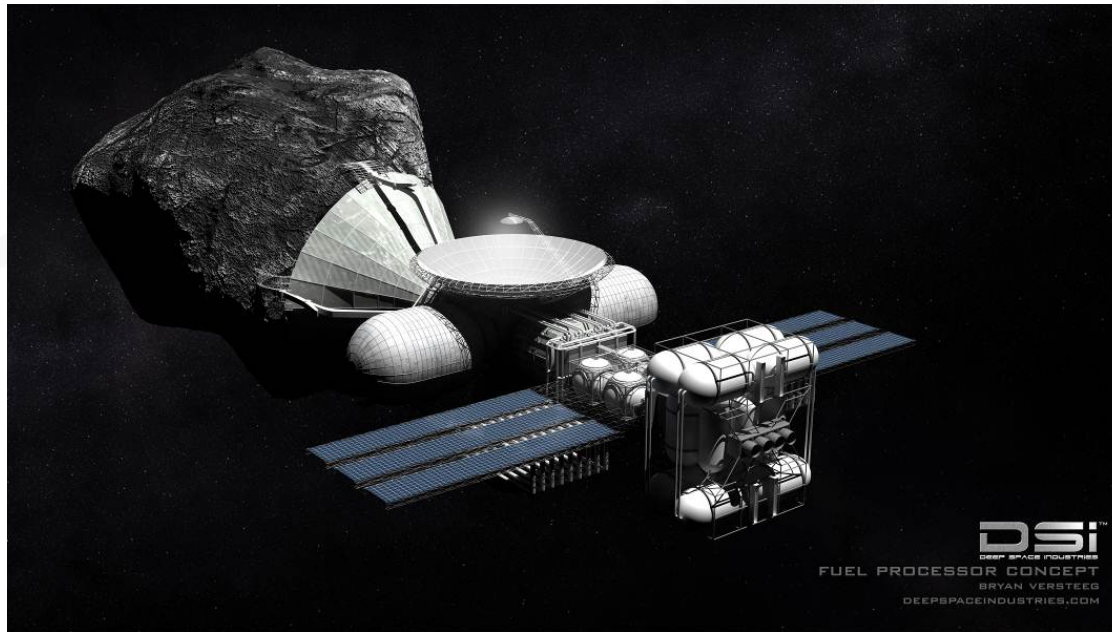


Aft View of ARRV

Hall Thrusters

## Asteroid Resource Utilization

- Accessibility to asteroid material in cislunar space allows in-situ resource utilization (ISRU) demonstrations to determine if material can be converted into useful products
  - Benefits future human exploration of the solar system
  - May be critical in advancing commercial efforts to mine asteroids



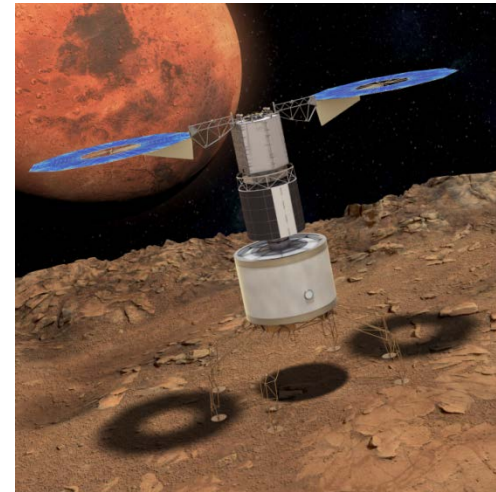
Asteroid Propellant Processing Concept

## Robotic Boulder Capture (RBC)

- Use of robotic arms to interact with an asteroid is applicable to other NASA missions
  - Satellite servicing and refueling
  - Providing surface contact and manipulation for a future crewed mission to NEAs or Martian moons
- End-effector/grippers (i.e., microspine technology) can be used for:
  - Future robotic exploration (lunar, Mars, Mars moons)
  - Exploration of bodies like NEAs, Phobos, Deimos, and main belt asteroids like Ceres



Hybrid Capture System Concept for the ARRM RBC Option

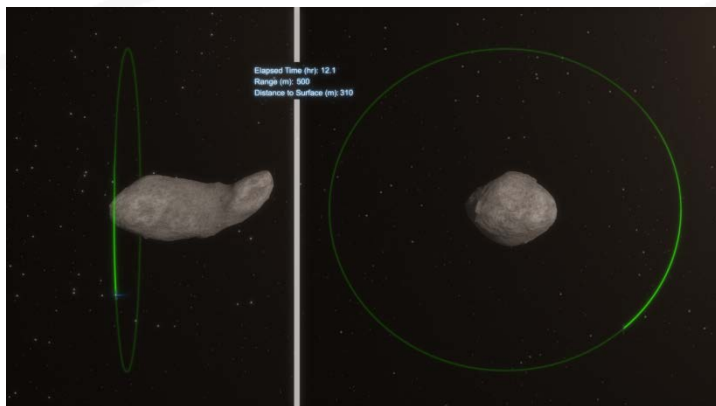


Spaceframe Capture Arms integrated with a Habitat on Phobos Surface

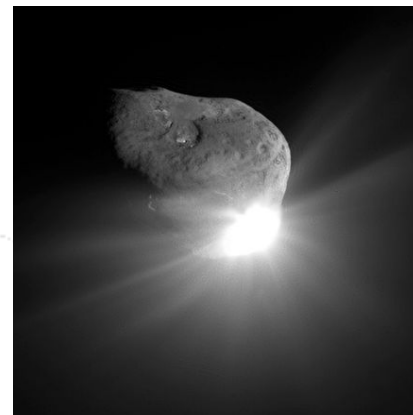


## Planetary Defense

- Planetary Defense techniques and experience possible through the Robotic Boulder Capture (RBC) option are relevant and extensible to future missions
  - Demonstrations performed on large NEA
    - Potential demonstrations include Enhanced Gravity Tractor (EGT), Ion Beam Deflection (IBD) and kinetic impact approach
- Current RBC option mission timeline includes EGT technique demonstration
  - Applicable to diverting NEAs from Earth-impacting trajectories given sufficient warning time
- Kinetic impact demonstration could be conducted and effectiveness compared to EGT



Demonstration of Enhanced Gravity Tractor on a Large NEA



Kinetic Impact on Comet 9P/Tempel by the Deep Impact Mission

# Asteroid Redirect Crewed Mission Extensibility

## Common Automated Rendezvous and Docking/Capture Sensors

- NASA needs automated rendezvous and docking/capture (AR&D) sensors for multiple missions
- NASA is pursuing a common suite of AR&D sensors to apply across multiple AR&D missions starting with the asteroid missions
  - Visible cameras paired with selectable lenses per mission needs
    - Medium resolution (narrow angle)
    - High resolution (wide angle)
  - 3D LIDAR
  - Infrared camera
- NASA created a common specification addressing environment and performance for each sensor which will fulfill each mission's AR&D needs
- NASA released a BAA to pursue sensors meeting this spec



Long Range	Medium Range	Close Range	Application of Common Suite
S-Band Transponder for Range/Range Rate to reduce timeline; Star Tracker for bearing High Resolution Camera for bearing		3D LIDAR for precise alignment for docking High Resolution Camera for secondary pose	High Resolution Camera 3D LIDAR
	Medium Resolution Camera for asteroid acquisition, spin rate and bearing to the asteroid	3D LIDAR for asteroid characterization and alignment for bag capture	Medium Resolution Camera 3D LIDAR
Medium Resolution Camera for bearing to the asteroid	Medium and High Resolution Cameras for spin rate, 3D map of the surface and boulder identification	3D LIDAR for 3D range images to the target boulder Medium and High Resolution Cameras for spacecraft pose and images of boulder collection areas	High Resolution Camera Medium Resolution Camera 3D LIDAR

\* Addition of infrared camera on asteroid missions for robustness is being assessed

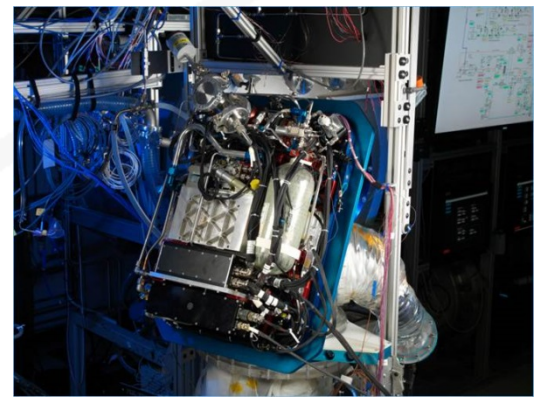
# Asteroid Redirect Crewed Mission Extensibility (Cont.)

## Extravehicular Activity (EVA)

- Aspects of the EVA segment within ARCM that are extensible to other missions include:
  - Providing contingency EVA capability to all manned Orion missions
    - Use of MACES (Orion LEA suit) with mobility enhancements
  - Exploration Portable Life Support System (PLSS) technologies applicable to exploration missions
    - Rapid Cycle Amine (RCA) swing bed for carbon dioxide removal
      - Vacuum missions only; addition of sweep gas required for use on Mars surface
    - Suit Water Membrane Evaporator (SWME) for thermal conditioning
  - Exploration PLSS combined with Z suit will be the EVA suit for all future human exploration missions
  - EVA tools and worksite stability techniques are extensible to any microgravity EVA including moons of Mars



Mobility Enhancements Testing of MACES in the NBL



PLSS 2.0 Test Unit on the Space Suit Assembly Simulator



Z2 Suit with PLSS

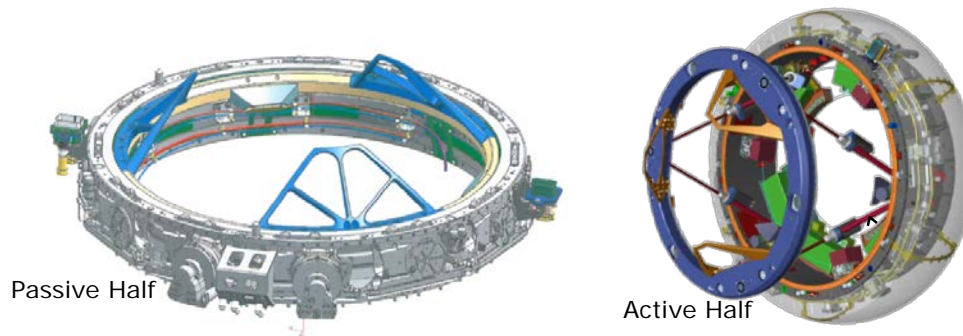


## Sample Collection

- Sample collection tools/techniques developed for ARCM that are extensible to future NASA missions include:
  - Collection of macroscopic samples from various locations on surface and/or interior
  - Sampling operations on small, airless body under low-gravity conditions
  - Methods to store and maintain samples during return journey for subsequent laboratory analyses on Earth
- Lessons learned from ARM can be applied to EVA planning for sample site assessment, documentation, sampling operations, and storage/containment activities on Phobos/Deimos

## Automated Docking

- Use of International Docking System Standard (IDSS)-compliant NASA Docking System (NDS) as standard docking system for all spacecraft allows delivery of different assets in separate flights that can be integrated later during the mission
  - Enables missions to Mars vicinity (Phobos and/or Deimos)
- Power and data transfer capabilities supports integrated stack control and power sharing

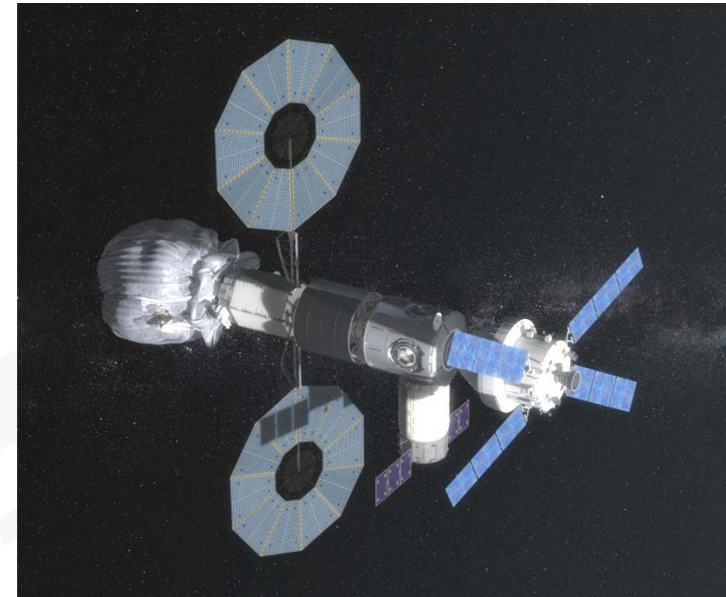


IDSS-Compliant NASA Docking System; passive half on ARRV, and active half on Orion



## Extended Crewed Missions to Asteroid in DRO

- Many possible opportunities for further utilization of the Asteroid
  - Testing of anchoring techniques
  - In-situ Resource Utilization (ISRU) Demonstration
  - Additional Asteroid Sample Collection
  - Lunar and Mars sample return
  - Scientific Experiments
  - Many other possibilities
- Addition of utilization elements provide:
  - Extended crewed mission duration and additional EVA capability
  - Enhance crew safety with more robust systems and infrastructure
  - Testbed for crew operations and systems required to sustain life for longer deep-space missions

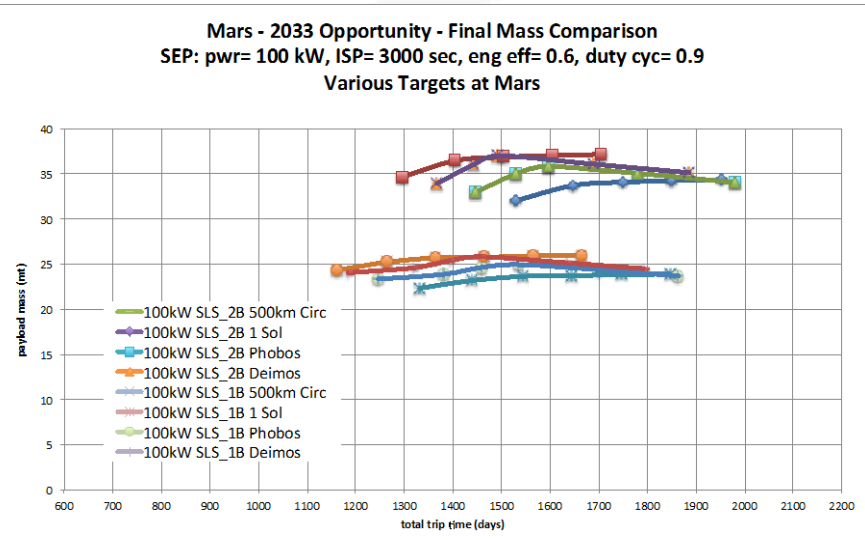
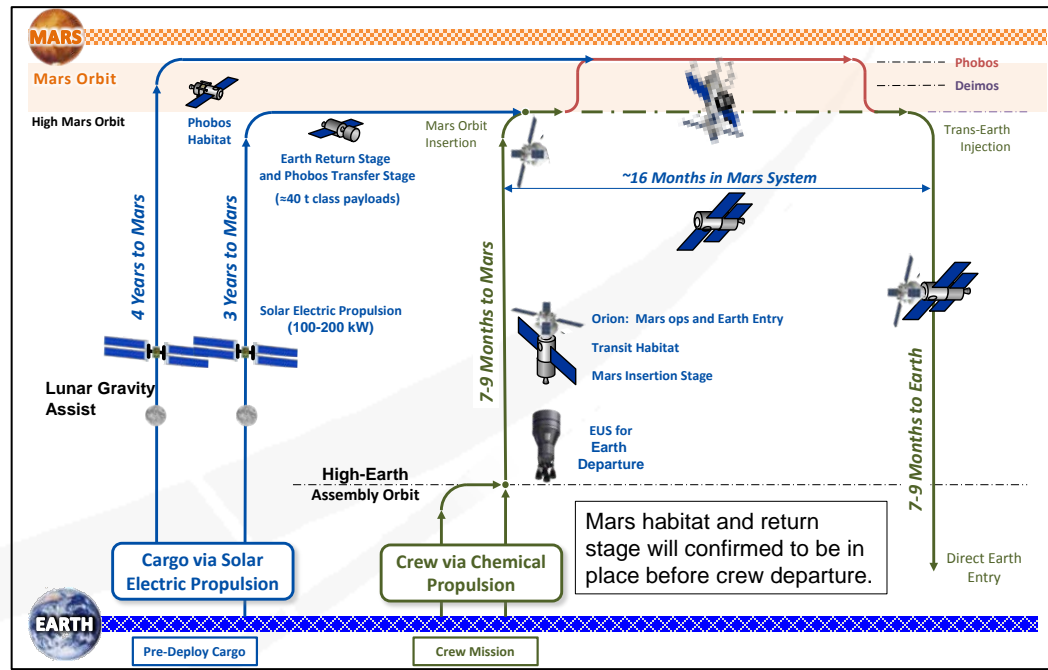


Notional Example of Extensibility of Asteroid Mission

# Extensibility to Missions Post-ARCM (Cont.)

## Mars Phobos Mission

- ARV-derived SEP requires “Split-Sprint” mission approach
- SEP used to pre-deploy crew habitat, Phobos systems, and return propulsion stage to Mars orbit
- Crew sent on “one-way” mission to Mars orbit via high-thrust chemical propulsion (critical rendezvous at Mars)
- Crew utilizes pre-deployed assets for Phobos exploration and eventual return to Earth



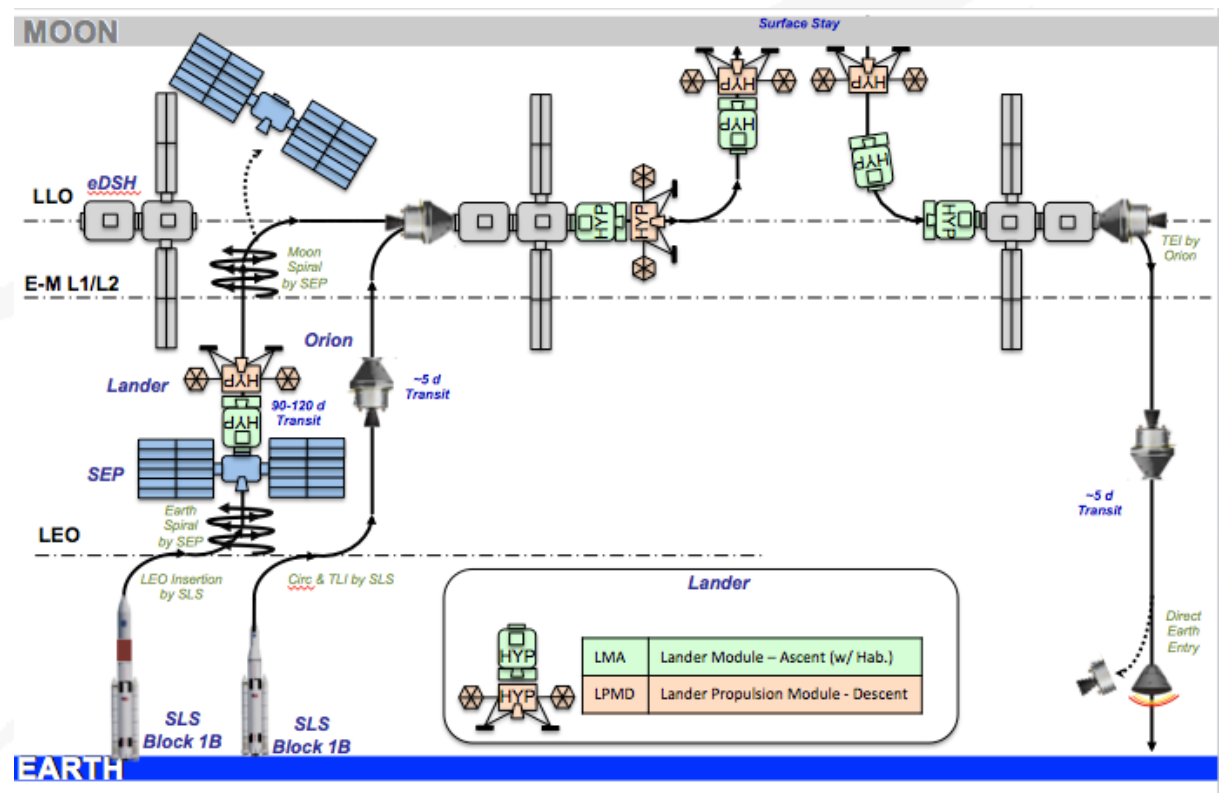
## ARRV-Derived SEP Cargo Tug

- Performance estimates indicate that ARRV-derived SEP may be sufficient for Phobos exploration
- Further assessments required to confirm performance including determining how this type of strategy can feed forward to future landed missions (heavier cargo elements such as Mars landers)

# Extensibility to Missions Post-ARCM (Cont.)

## Lunar Mission

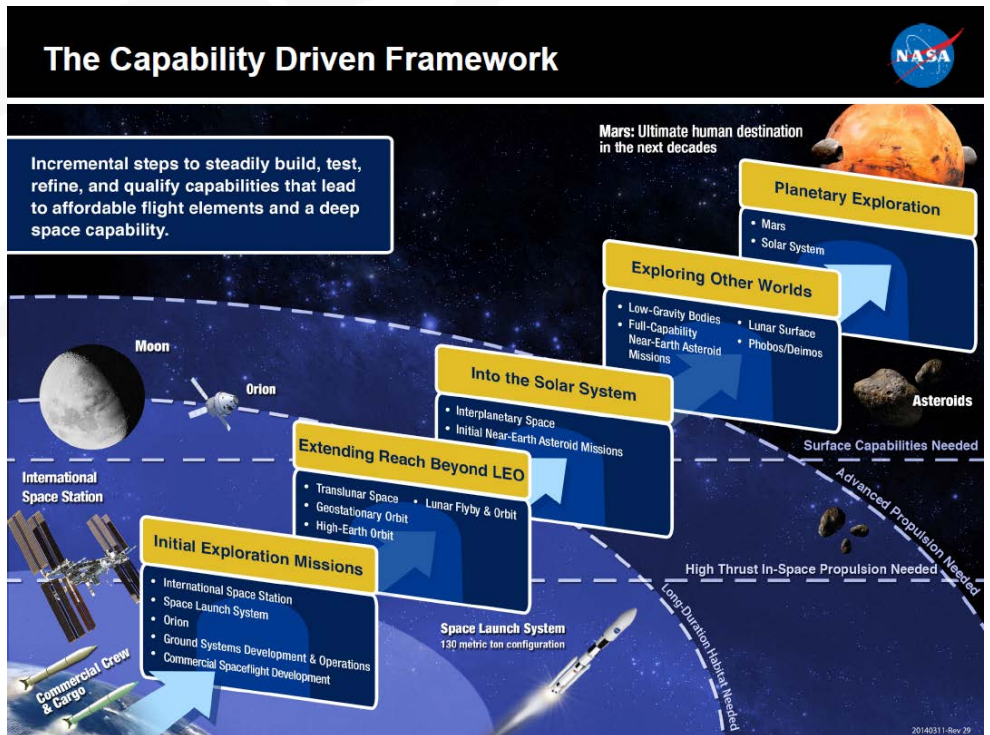
- ARM operational experience and systems could be leveraged to support exploration of the lunar surface
  - Delivery of uncrewed assets to lunar vicinity using SEP system, allowing for additional payload/mission capabilities
  - SEP bus could be derived from ARRV



# Extensibility to Missions Post-ARCM (Cont.)

## Capability Driven Framework (CDF)

- ARM technologies support the CDF as they enable implementation of many possible missions
  - ARM SEP system could be used to support lunar missions
    - Power level could be evolved to preposition assets for Mars missions
  - AR&D sensor suite directly support lunar, Mars, and other possible missions
  - Orion and SLS systems enable “extending reach beyond LEO”
  - Provides operational experience working on low gravity bodies (“exploring other worlds”)



- The Asteroid Redirect Mission has great promise for extensibility
  - SEP, AR&D sensors, and RBC Capture mission are all extensible to possible lunar or Mars moon missions
  - Docking and EVA systems generically support all human exploration missions
- Investment in ARM concept provides logical stepping stones to future exploration goals





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