

# NASA's Space Launch System: An Enabling Capability for Discovery

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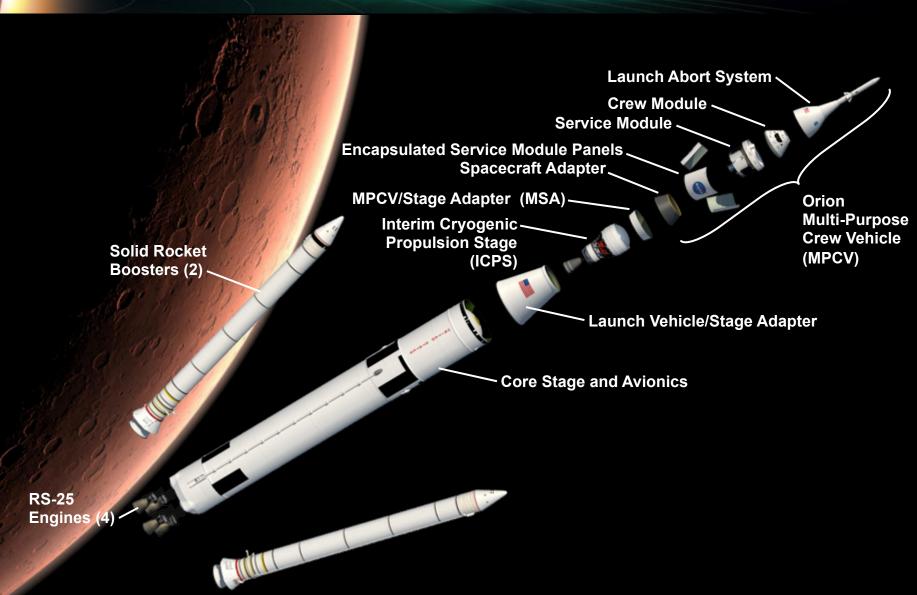




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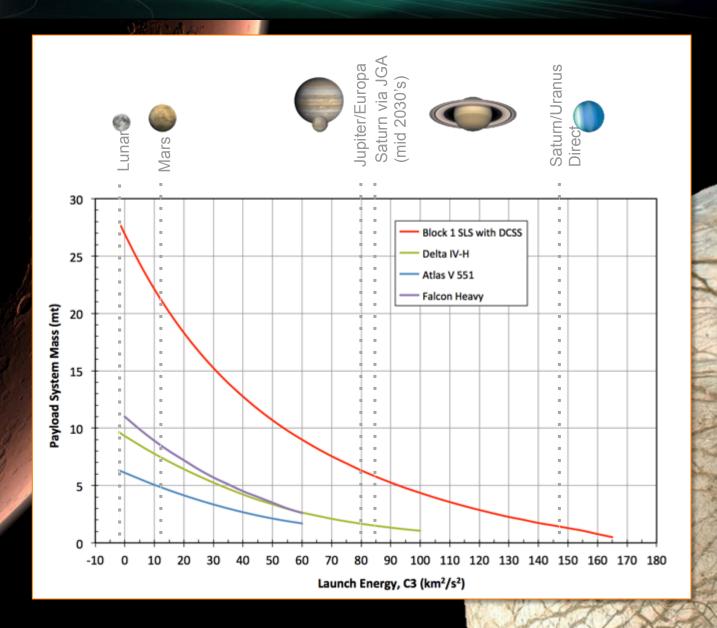
# **SLS Initial Configuration**





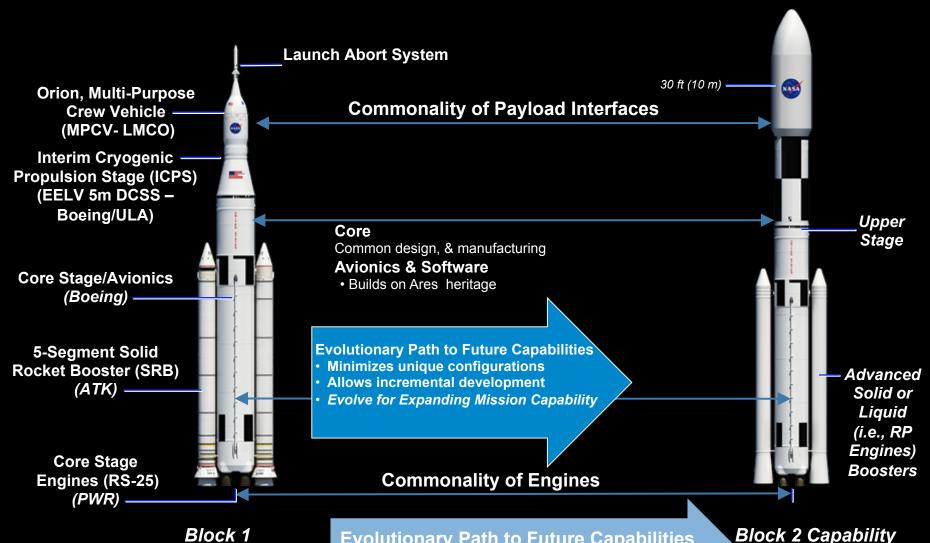
# **SLS Initial Configuration Performance**





# **SLS Block Commonality**





Initial Capability, 2017-21 70 metric ton Payload

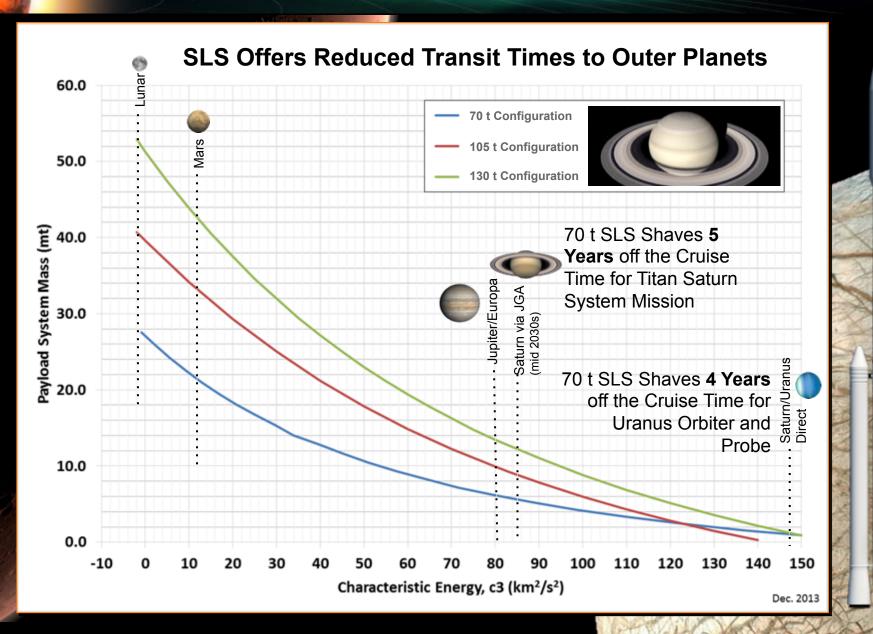
**Evolutionary Path to Future Capabilities** 

- Minimizes unique configurations
- Allows incremental development

130 metric ton **Payload** 

### **SLS Evolved Performance**





# **SLS Offers Unrivaled Payload Volume**



 SLS is investigating utilizing existing fairings for early cargo flights, offering payload envelope compatibility with design for current EELVs

Phase A studies in work for 8.4m and10 m fairing options



4m x 12m (100 m<sup>3</sup>)



5m x 14m (200 m<sup>3</sup>)



5m x 19m (300 m<sup>3</sup>)



8.4m x 31m (1200 m<sup>3</sup>)

**10m x 31m** (1800 m<sup>3</sup>)

## **Global Exploration Roadmap**

























2013 2020 2030



#### **International Space Station**

General Research and Exploration **Preparatory Activities** 

Note: ISS partner agencies have agreed to use the ISS until at least 2020.

Commercial or Government Low-Earth Orbit Platforms and Missions

#### **Robotic Missions to Discover and Prepare**



Mars Sample Return and Precursor **Opportunities** 

#### **Human Missions Beyond Low-Earth Orbit**

Explore Near-Earth Asteroid



**Extended Duration Crew** Missions

> Humans to Lunar Surface

Missions to Deep Space and Mars System



Sustainable **Human Missions** to Mars Surface

# **SLS Mission Capabilities**



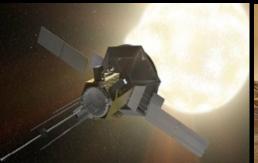


**Space Habitat** 

**Asteroid Rendezvous** 

**Deep Space Telescope** 

**GEO Servicing** 







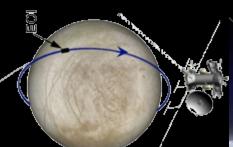


**Solar Probe** 

**Mars Sample Return** 

**Mars Cargo Lander** 

**Humans to Mars** 











**Europa Clipper** 

**Enceladus Return** 

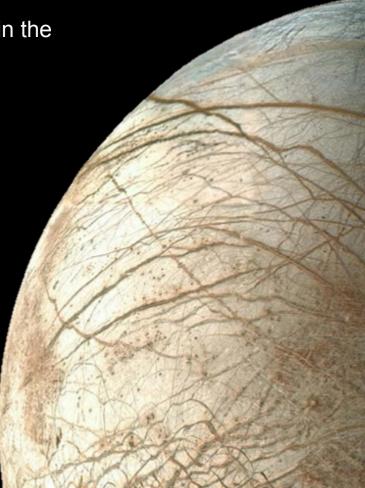
**Uranus Spacecraft** 

Interstellar

### **SLS Benefits for Science Missions**

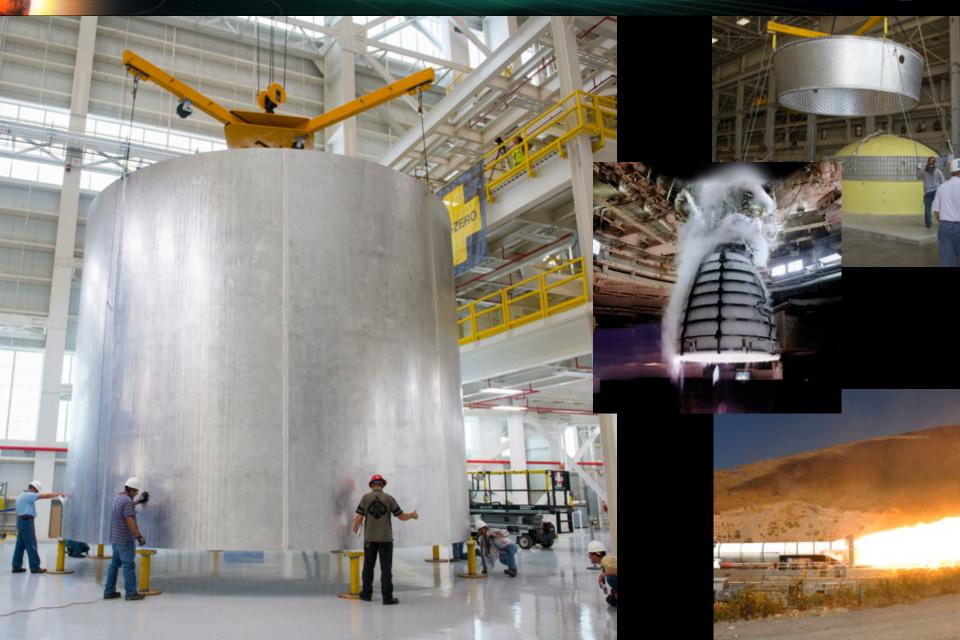


- SLS Being Developed to Enable Exploration
  - Volume and mass capability/margin required for complex deep-space human mission
    - -Increased design simplicity
    - Fewer origami-type payload designs needed to fit in the fairing
    - -Simplifies on-orbit operations
    - -Reduced risks and hazards
- SLS investment can be leveraged for other missions requiring large volume or mass, or reduced trip times
  - Deep Space Exploration
  - -Planetary Landers
  - -Human Habitats
  - -Great Observatories
  - -Space Solar Power
  - -Outer Planet Missions
  - -National Security Space Payloads



# Manufacture and Testing Underway





# Summary



#### SLS provides capability for human exploration missions.

- 70 t configuration enables EM-1 and EM-2 flight tests.
- Evolved configurations enable missions including humans to Mars.

#### SLS offers unrivaled benefits for a variety of missions.

- 70 t provides greater mass lift than any contemporary launch vehicle; 130 t offers greater lift than any launch vehicle ever.
- With 8.4m and 10m fairings, SLS will over greater volume lift capability than any other vehicle.
- Initial ICPS configuration and future evolution will offer high C3 for beyond-Earth missions.

#### SLS is currently on schedule for first launch in December 2017.

- Preliminary design completed in July 2013; SLS is now in implementation.
- Manufacture and testing are currently underway.
- Hardware now exists representing all SLS elements.