National Aeronautics and Space Administration

# Space Launch System And NASA's Path to Mars

NASA

Todd May Program Manager Space Launch System (SLS) Program *August 2014* 



www.nasa.gov/sls

#### lars Is The Mission



launch capabilities to improve responsiveness, resiliency, and cost effectiveness for future space

In support of civil space programs and activities, including human and robotic spaceflight for exploration, scientific, operational, and other civil purposes, the Administrator of NASA shall:

Develop, in support of U.S. space exploration goals, the transportation-related capabilities necessary to support human and robotic exploration to multiple destinations beyond low-Earth TITLE orbit, including an asteroid and Mars. Such capabilities include a heavy-lift space transportation system, crew vehicles, and other related capabilities such as in-space refueling technologies and **SPAC** NATIONAL SPAUL ~--EARTH ORBIT 42 USC 18321.

#### SEC. 301. HUMAN SPACE FLIGHT BEYOND LOW-EARTH ORBIT.

(a) FINDINGS.—Congress makes the following findings: (1) The extension of the human presence from low-Earth orbit to other regions of space beyond low-Earth orbit will enable missions to the surface of the Moon and missions to deep space destinations such as near-Earth asteroids and Mars. (2) The regions of cis-lunar space are accessible to other national and commercial launch capabilities, and such access raises a host of national security concerns and economic

2014 Strategic Plan

space frontier for the ISS one-of-a-kind exploration missions use of the unique micro-

academic science and research. We will work with to ensure a strong U.S. capability for launching crew and cargo into space. We will complete development of next-generation space systems like the Space Launch System (SLS) and the Orion Multi-Purpose Crew Vehicle to take us past low Earth orbit, and set a pathway to Mars and beyond. We will deploy the James Webb Space Telescope (JWST) to glimpse back in time to the formation of the first stars and galaxies, while our New Horizons mission will uncover knowledge

# Mars: The Human Destination



### he Path to Mars



#### **EARTH RELIANT** MISSION: 6 TO 12 MONTHS RETURN TO EARTH: HOURS

#### PROVING GROUND MISSION: 1 TO 12 MONTHS RETURN TO EARTH: DAYS

#### MARS READY MISSION: 2 TO 3 YEARS RETURN TO EARTH: MONTHS

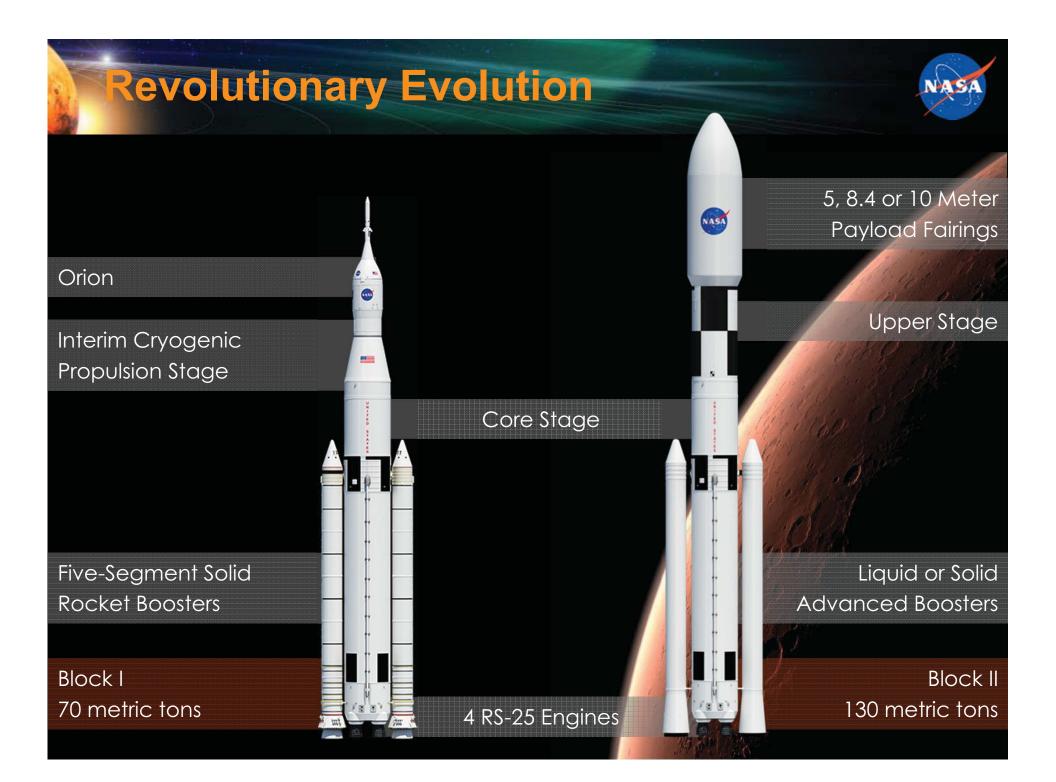
Mastering fundamentals aboard the International Space Station

U.S. companies provide access to low-Earth orbit Expanding capabilities by visiting an asteroid redirected to a lunar distant retrograde orbit

The next step: traveling beyond low-Earth orbit with the Space Launch System rocket and Orion spacecraft Devel by ex

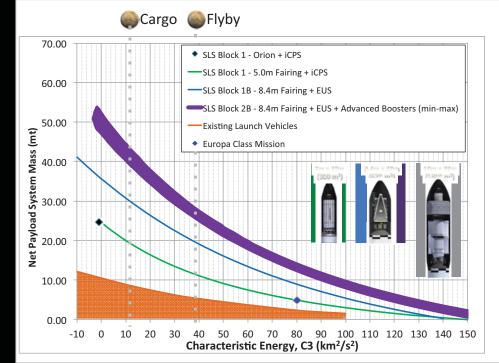
Developing planetary independence by exploring Mars, its moons and other deep space destinations



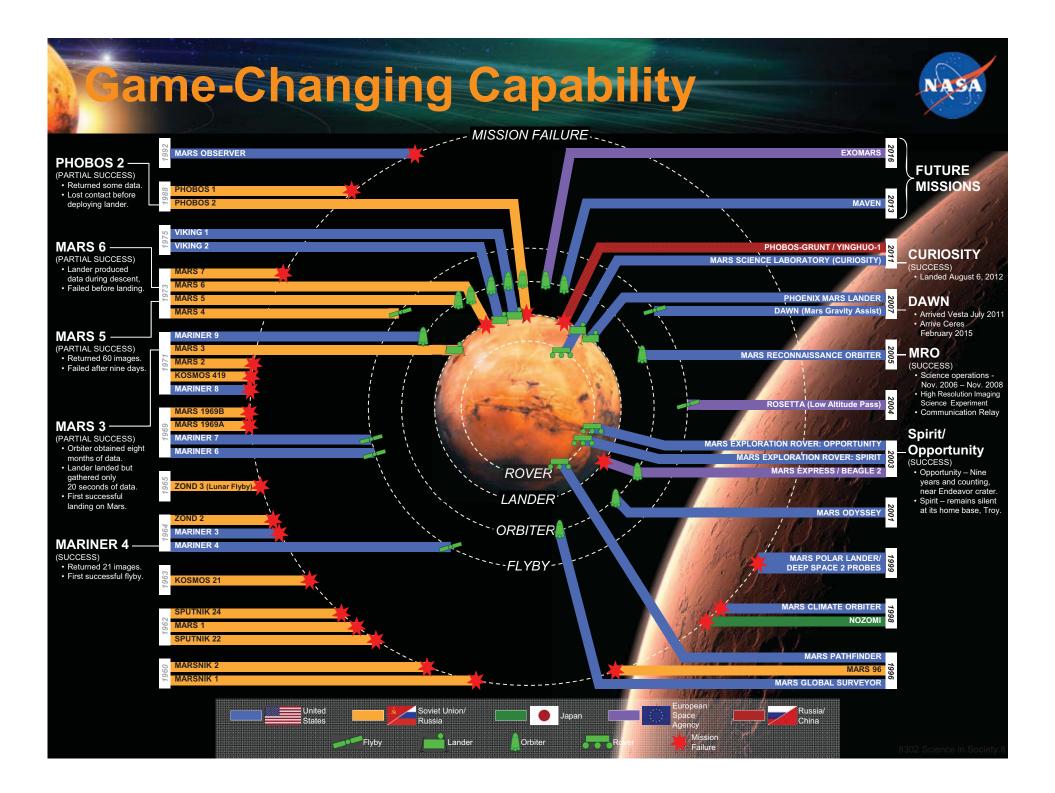


### **Benefit: High Departure Energy**

- Even the Initial configuration of SLS offers orders of magnitude greater payload-to-destination energy compared to existing launch vehicles; future configurations improve C3 performance even further.
- Higher departure energy offers more launch opportunities.
- Trade space exists between departure energy and mass capability.





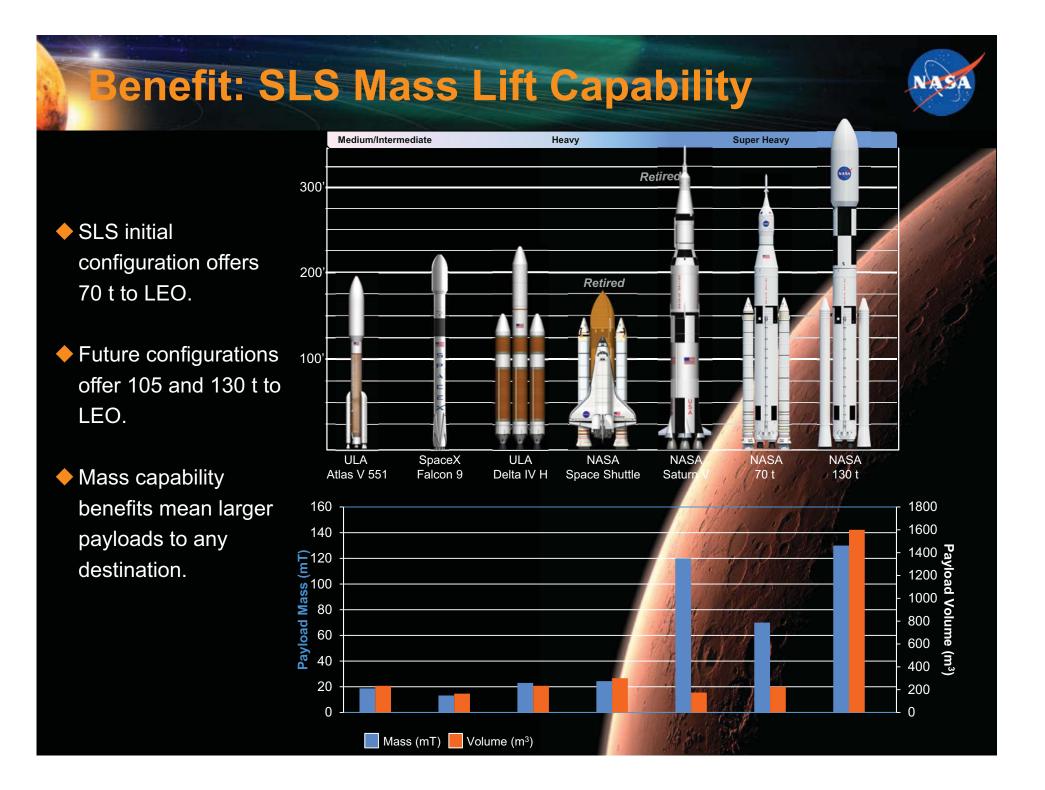


## Case Study: Mars Fly-By

NASA

500-600-day free-return fly-by of Mars; 2021 window would include fly-by of Venus.

SLS is uniquely capable of providing the required mass-todeparture-energy needed for the mission.



#### **Benefit: Unrivaled Payload Volume**

5m x 19m

 $(300 \text{ m}^3)$ 

8.4m x 31m

 $(1200 \text{ m}^3)$ 

10m x 31m

 $(1800 \text{ m}^3)$ 

- SLS is investigating utilizing existing fairings for early cargo flights, offering payload envelope compatibility with design for current EELVs
- Phase A studies in work for 8.4m and 10 m fairing options

5m x 14m

 $(200 \text{ m}^3)$ 

4m x 12m

(100 m<sup>3</sup>)

#### Case Study: Mars Sample Return



- Robotic precursor mission to return material samples from Mars
- 70-t SLS could support onelaunch Mars Sample return mission with sampling rover carrying ascent vehicle and inspace return vehicle
- SLS could support two-flight sample return in conjunction with 2020 Mars science rover mission, launching ascent vehicle and inspace return vehicle

### ase Study: Martian Moons

- Provides stepping-stone opportunity toward human landing on Mars
- Allows for real-time humanrobotic telepresence exploration of Martian surface
- Provides ambitious Mars-vicinity target while developing Martian EDL capability
- SLS offers mass and volume capability for needed habitation and propulsion systems

#### Case Study: NASA DRA5

6)

NASA's 2009 Design Reference Architecture 5 outlines plan for 900-day human mission to Mars

DRA5 requires 825 metric tons initial mass to low Earth orbit (almost double ISS launch mass).

 SLS exceeds DRA5 minimum 125+ t to LEO mass launch requirement

 SLS meets DRA5 minimum 10-mdiameter fairing requirement

 SLS enables crewed mission to Mars as outlined by DRA5

Source: "Human Exploration of Mars Design Reference Architecture 5.0," NASA Mars Architecture Steering Group, July 2009

#### **Recent Progress**

Launch Vehicle Stage Adapter: Contract awarded in February 2014.

Avionics: Avionics "first light" marked in January 2014; currently testing most powerful flight system computer processor ever.



**Boosters:** Forward Skirt test completed May 2014; preparations underway for QM-1.







----

-

First flight hardware currently in Florida for Exploration Flight Test-1 in Fall 2014.

**Core Stage:** Initial confidence barrels and domes completed; Vertical Assembly Center installation to



be completed in July 2014.







**Engines:** First RS-25 engine fitted to A-1 stand at Stennis Space Center; testing begins October 2014.

# Building to Exploration Mission-1 (EM-1) NASA

09/2011 Tested Booster Development Motor

S

ENT

∑ H

LIS

0 M P

ັ ບ

∢

WHAT'S NEXT

07/2012 Delivered RS-25 Engines to Inventory

07/2013 Competed Preliminary Design Review

10/2011 - 12/2013 Tested SLS Wind Tunnel Models

07/2013 Completed First Confidence Barrel Section Welding

10/2013 Completed Thrust Vector Control Test

11/2013 Conducted Adaptive Augmenting Control Flight Test

12/2013 Completed LOX Forward Dome Manufacturing Demo

1/2014 Conducted Avionics "First Light" in Integration Facility

02/2014 Shipped Multi-Purpose Crew Vehicle Stage Adapter for EFT-1

07/2014 Complete Manufacturing Tooling Installation

07/2014-15 Test Main Engines, Boosters , & Core Stage Structure

7/2015 Complete the SLS Critical Design Review

6/2016 Assemble the Core Stage Assembly and Test Fire

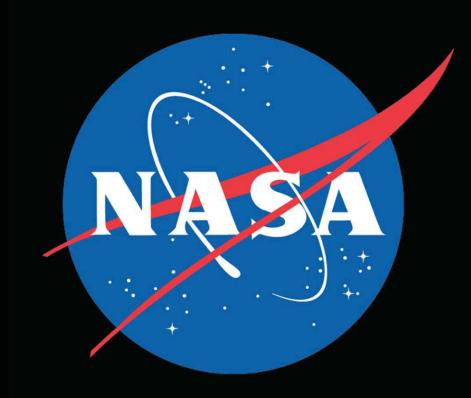
7/2017 Stack the SLS Vehicle

12/2017 Transport SLS from the VAB to the Launch Pad









Man cannot discover **new oceans** unless he has the **courage to lose sight of the shore.** 

**For More Information** 

www.nasa.gov/sls

www.twitter.com/nasa sls

www.facebook.com/nasasls