



Functional Performance of an Enabling Atmosphere Revitalization Subsystem Architecture for Deep Space Exploration Missions

Jay Perry, Morgan Abney, Kenneth Frederick, Zachary Greenwood, Matthew Kayatin, Robert Newton, Keith Parrish, Monsi Roman, and Kevin Takada — NASA Marshall Space Flight Center

Lee Miller — ECLS Technologies, LLC

Joseph Scott — Jacobs Engineering

Christine Stanley — Qualis Corporation

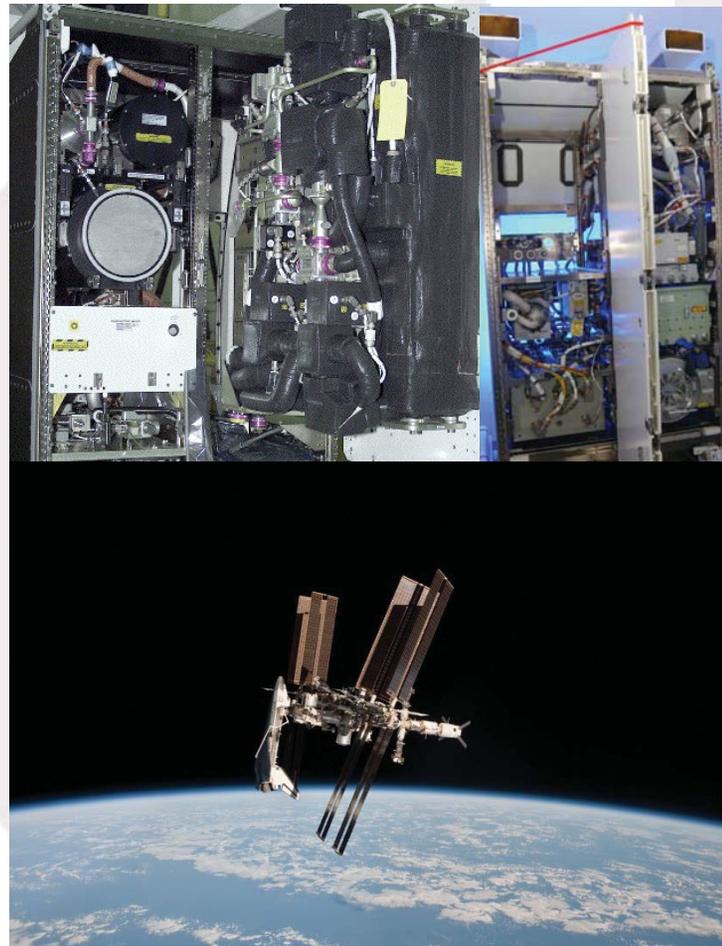
43rd ICES, 15–18 July 2013, Vail, Colorado, USA

AIAA-2013-3421



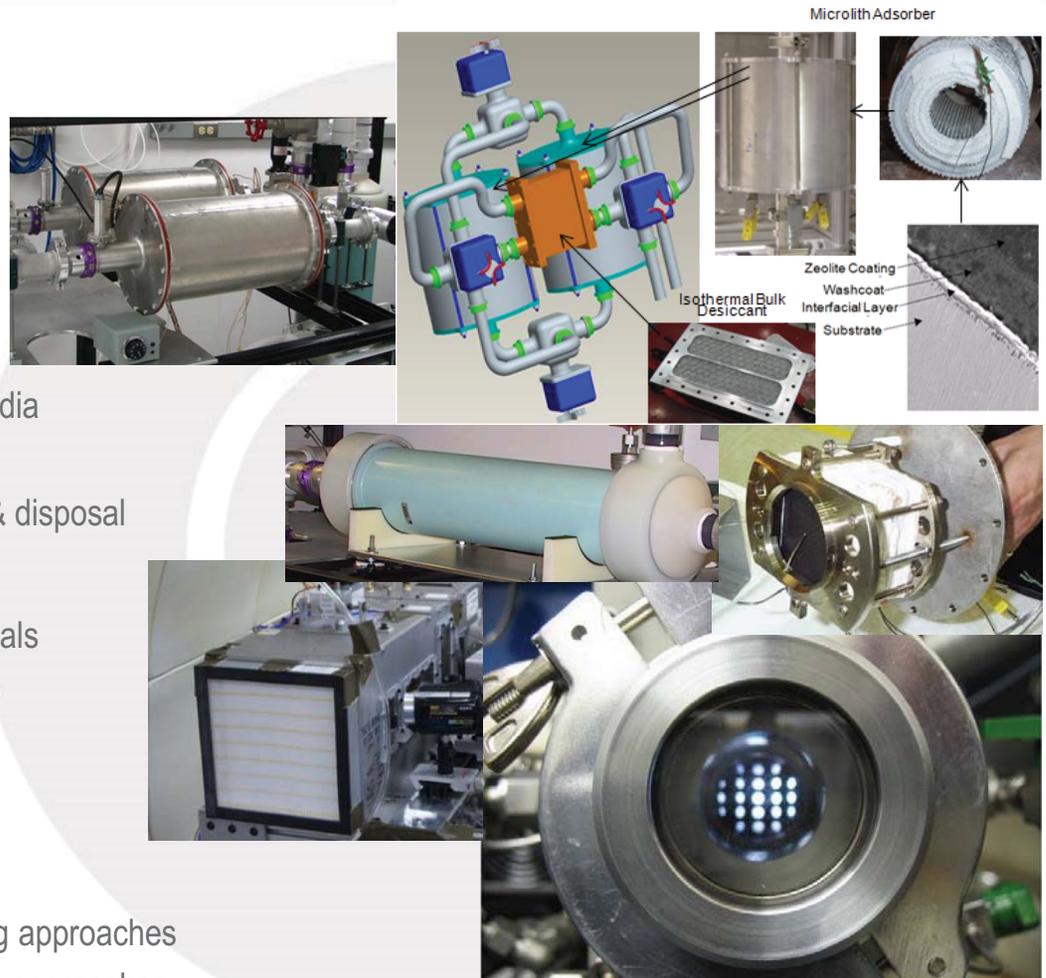
Technological Advancement Objectives

- To demonstrate an evolved ISS atmosphere revitalization (AR) subsystem architecture to enable deep space exploration
 - Improve reliability & maintainability
 - Reduce consumable mass
- To maximize commonality across missions and vehicles
- To mature process technologies for flight programs
 - Reduce technical risk and cost
- To develop modular resource recovery technologies



Strategic Improvements

- Cabin ventilation
 - Quiet fan design principles
- Carbon dioxide removal
 - Durable adsorbent media
 - Process air drying
- Trace contaminant control
 - Alternative high capacity adsorbent media
 - Structured oxidation catalysts
 - Low maintenance particulate filtration & disposal
- Oxygen supply
 - Long-lived electrolysis cell stack materials
 - Alternative process control approaches
- Oxygen recovery
 - Reduction byproduct processing
- Environmental monitoring
 - Alternative major constituent monitoring approaches
 - Alternative trace constituent monitoring approaches
 - Microbial & particulate monitoring techniques



Test Facility Overview

- **Characteristics**
 - 90.6 m³
 - Stainless steel
 - Vacuum-capable
- **Test support capabilities**
 - Metabolic simulation
 - Trace contaminant injection
 - Temperature and humidity control
 - Space vacuum resource simulation
 - Gas sample acquisition and analysis
 - Major constituent monitoring
 - Total pressure and atmosphere composition control
 - Process control & data acquisition/archiving
 - LabVIEW
 - Payloads and Components Real-time Automated Test System (PACRATS)
- **Analytical Instrumentation – VOC Analysis**
 - Agilent 5890 GC with a flame ionization detector coupled with a Markes TT24-7 Thermal Desorption System autosampler
 - Agilent 7890 GC with flame ionization and mass selective detectors coupled with a Gerstel Thermal Desorption System
 - MKS Multigas™ 2030 Fourier Transform Infrared Spectrometer
- **Analytical Instrumentation – Major Constituent Monitoring**
 - Oxigraf Model O2 analyzer – solid-state laser diode absorption
 - Sable Systems CA-2A analyzer – solid-state infrared absorption
 - Sable Systems RH-100 monitor – solid-state thin film capacitance



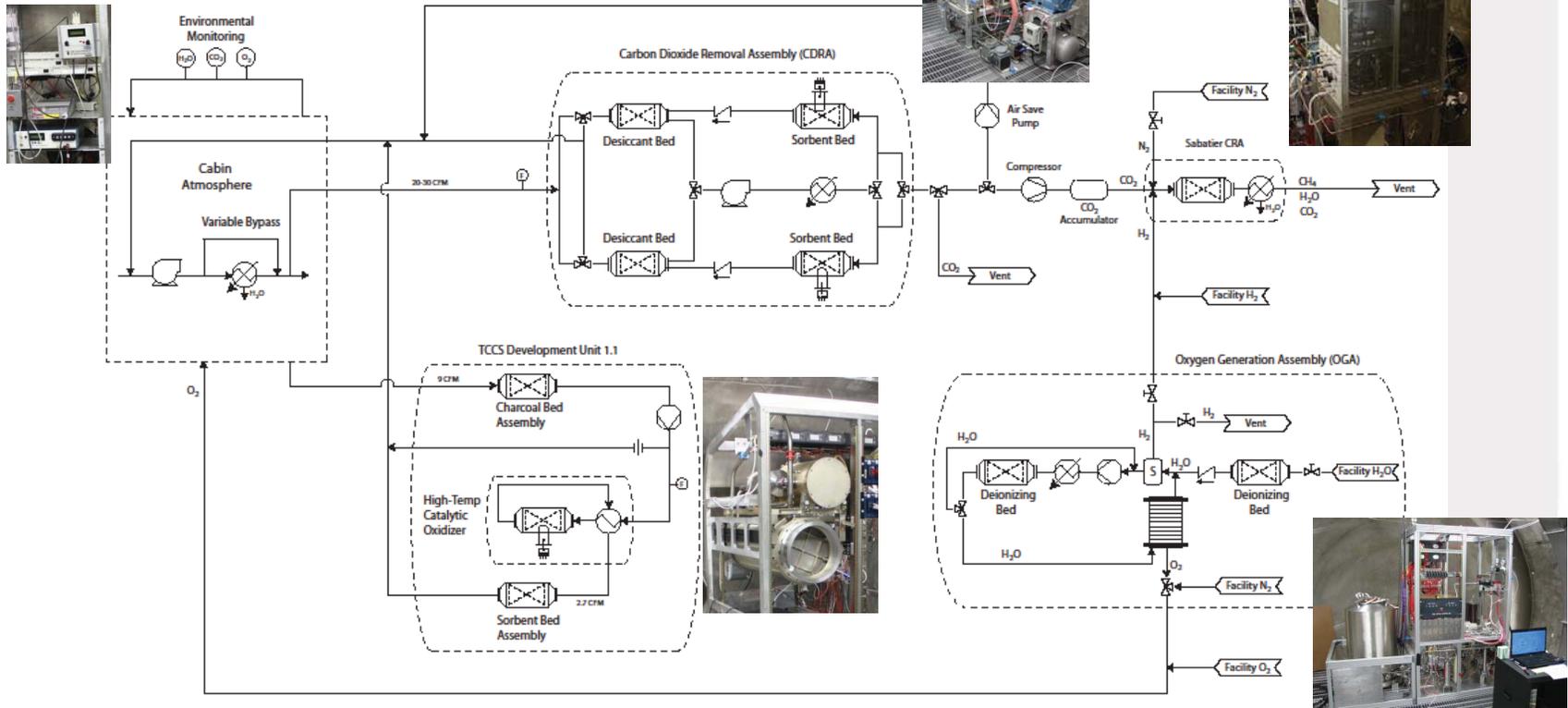
PARAMETER	RANGE
Total pressure	400-933 Pa gauge
Oxygen partial pressure	20.58±0.14 kPa
Carbon dioxide partial pressure	400±67 Pa
Temperature	21±2.8 °C
Relative humidity	50±5%
Contaminant injection rate	230 mg/hour*

*Percent by mass: methanol (10.7), ethanol (67.1), 2-propanol (4.8), ethanal (7.6), dimethylbenzene (2.3), dichloromethane (1.3), 2-propanone (6.2)

ISS Architecture Testing Objectives

- Demonstrate functional performance of the basic ISS AR subsystem using the CDRA in CO₂ vent mode and the TCCS operating in parallel.
- Demonstrate the partial functional performance of the basic ISS AR subsystem when operating in a resource recovery mode that includes integration with CO₂ conditioning, storage, and reduction equipment.
- Investigate propagation of trace contaminants through the core ISS AR subsystem equipment with emphasis on the CDRA and CO₂ conditioning and storage equipment.
- Demonstrate the full resource recovery functional performance of the ISS AR subsystem including the CO₂ removal, CO₂ conditioning and storage, CO₂ reduction and post-processing, oxygen generation, and trace contaminant control functions.

ISS Architecture Testing



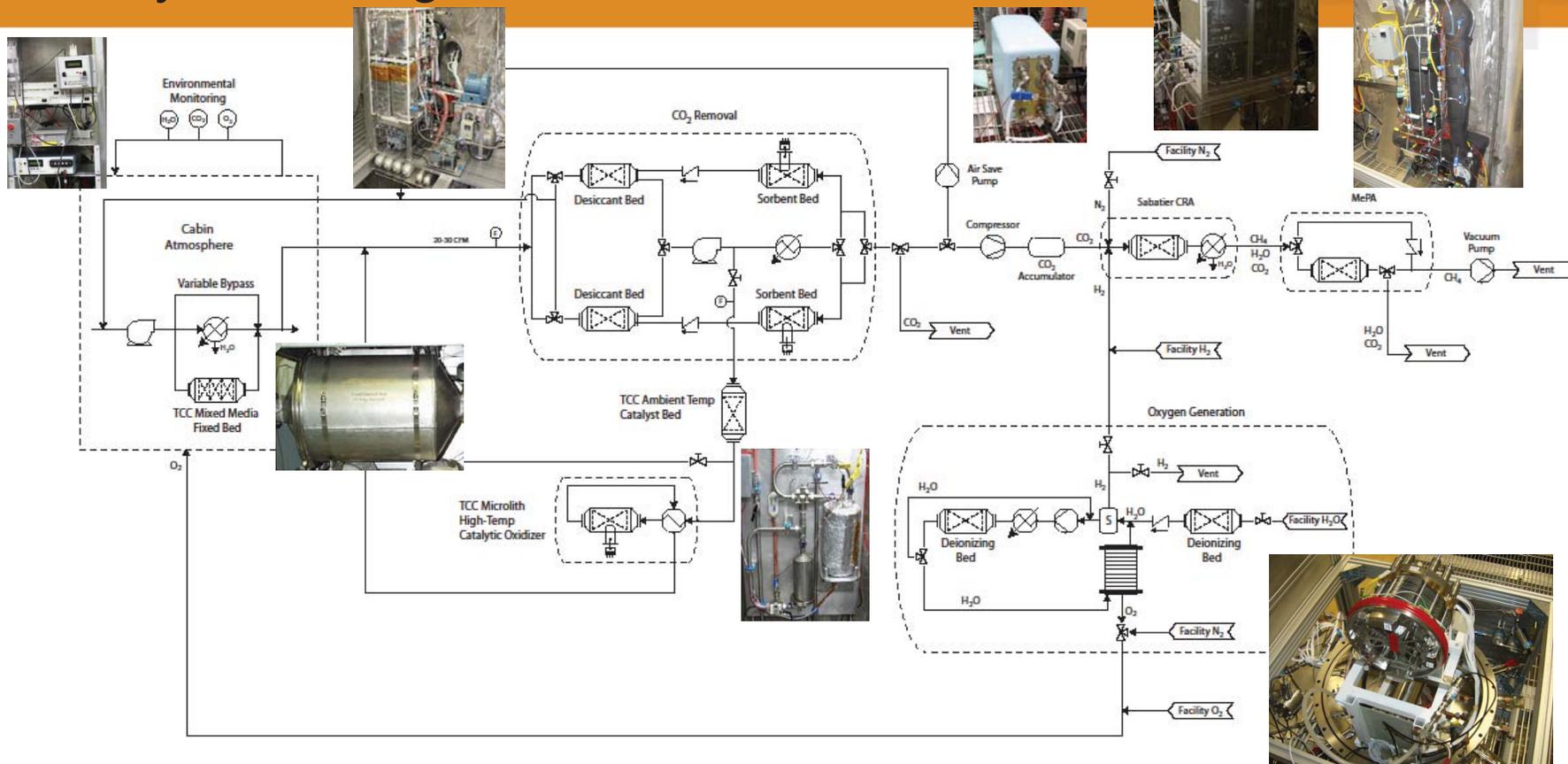
Symbols

	Packed bed		Check valve		Pump		Electrolysis Stack
	Heater		Three-way automatic control valve		Compressor		Accumulator
	Cooler		Two-way hand-operated valve		Blower		Separator
	Recuperative Heat exchanger		Dewpoint analyzer		Flowmeter		Orifice
	Condensing Heat exchanger		Carbon dioxide analyzer		Oxygen analyzer		

Cycle 1 Testing Objectives

- Demonstrate integrated modified ISS subsystem operation
 - Oxygen generation
 - CO₂ removal
 - Trace contaminant control
 - Major constituent monitoring
 - CO₂ reduction with partial post-processing first stage to purify methane
- Demonstrate CO₂ management compressor operation
 - Observe how CO₂ removal process valve valve sequencing affects overall CO₂ reclamation efficiency for various modes of operation.
- Determine process fluid purities
 - Product CO₂ from the CDRA-4 sorbent beds.
 - Product oxygen and hydrogen from the OGA.
 - Product water from the Sabatier-based CRA.
- Determine the effect cabin atmosphere leakage and/or atmospheric major constituent inclusion on the CDRA CO₂ product may have on CRA performance.

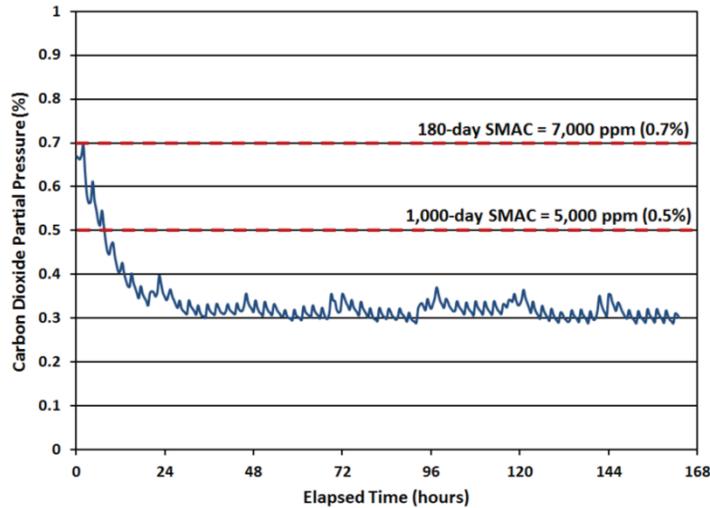
Cycle 1 Integrated Process Architecture



Symbols

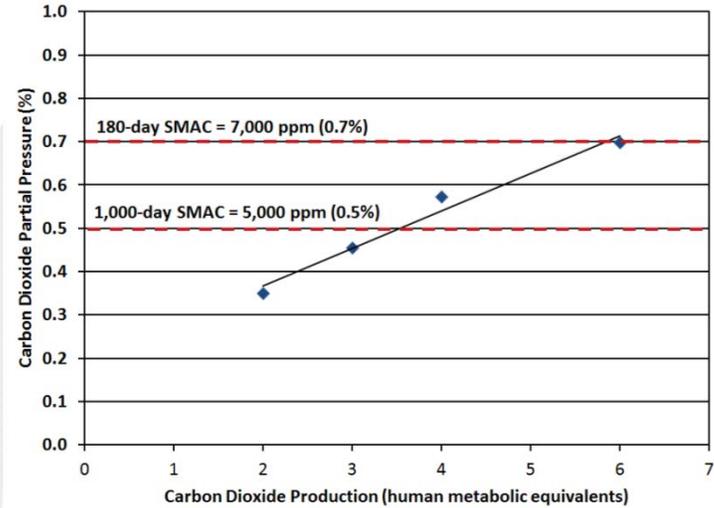
	Packed bed		Check valve		Pump		Electrolysis Stack
	Heater		Three-way automatic control valve		Compressor		Accumulator
	Cooler		Two-way hand-operated valve		Blower		Separator
	Recuperative Heat exchanger		Dewpoint analyzer		Flowmeter		Orifice
	Condensing Heat exchanger		Carbon dioxide analyzer		Oxygen analyzer		

Results: CO₂ Removal



ISS Architecture

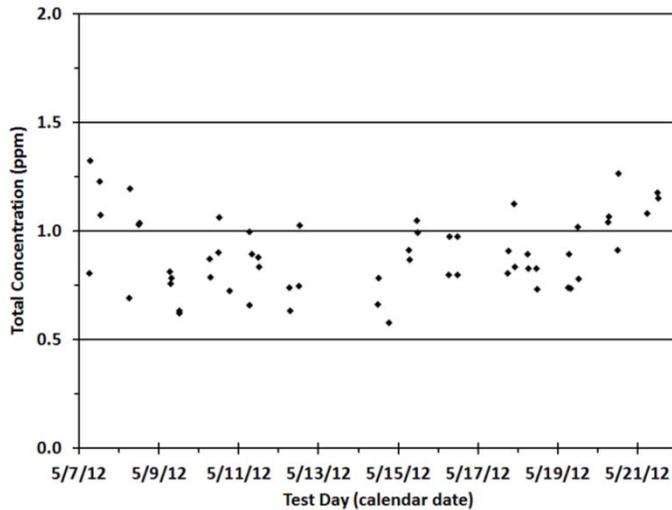
- ASRT zeolite 5A
- 34.6 m³/h flow
- 3-CM load
- 59% removal efficiency
- 0.34% CO₂ partial pressure



Cycle 1 Architecture

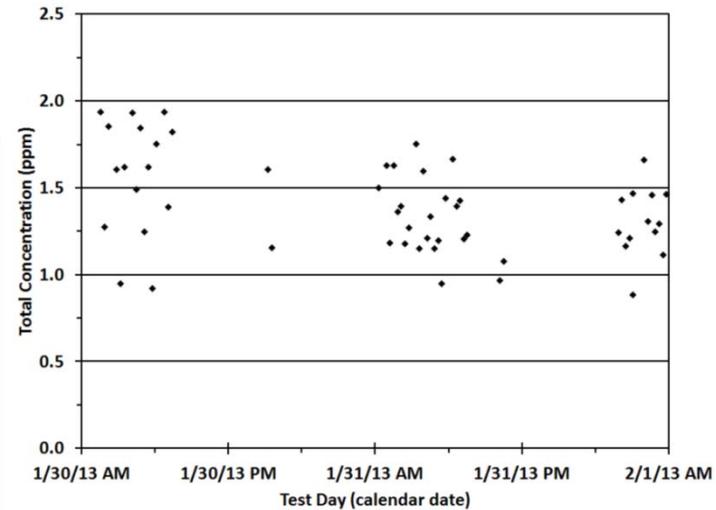
- RK-38 zeolite 5A
- 28.8 m³/hour flow
- 2-CM to 6-CM load
- 59% removal efficiency
- 0.45% CO₂ partial pressure

Results: Trace Contaminant Control



ISS Architecture

- Stand-alone TCC assembly
 - Carbon bed containing B-S Type 3032
 - Catalytic oxidizer containing Engelhard catalyst pellets
- 15.3 m³/h through carbon bed/4.6 m³/h through catalytic oxidizer
 - 88% mean single pass removal efficiency
- Indication of ethanol breakthrough of the carbon bed near end of testing
- 1.14 ppm mean total contaminant concentration

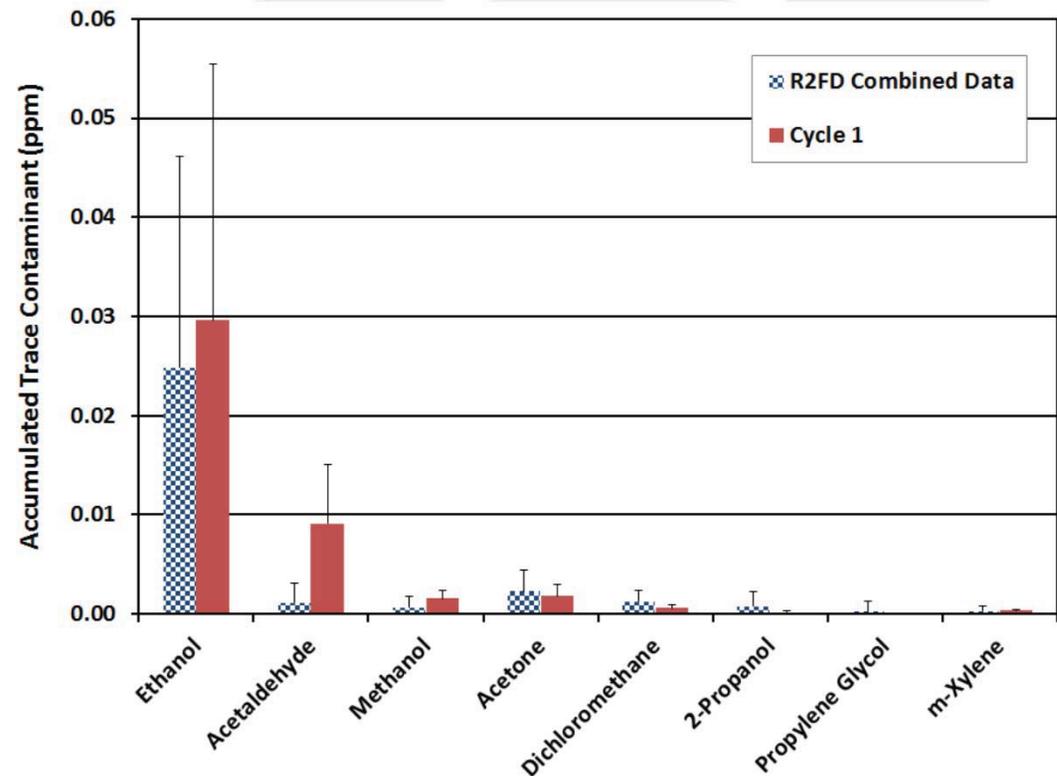


Cycle 1 Architecture

- Distributed TCC components
 - Carbon bed containing Chemsorb 1425 integrated with ventilation duct
 - Catalytic oxidizer containing Microlith™ catalytic reactor core integrated with CDRA
 - Eliminates electrical box and post-sorbent bed
- 8.5 m³/h through carbon bed
 - 91.2% mean single pass efficiency
- 5.9 m³/h through catalytic oxidizer
- 1.34 ppm mean total contaminant concentration

Results: CO₂ Reduction

- Water production rates consistent across architectures
 - ~2 ml/minute
- Consistent VOC loading in CO₂ feed
 - <0.03 ppm
- Suspected CO₂ reduction reaction poisons <<0.01 ppm
 - No indication of performance degradation during testing phases



Improvements for the Future

- Trace Contaminant Control
 - Full treatment of the ventilation flow with activated carbon cartridges mounted in main duct
- CO₂ Removal
 - Tune process parameters
 - Increase total flow by 6.6% to 37 m³/h to accommodate TCC catalytic oxidizer flow
 - Investigate half-cycle time and bed regeneration temperature modifications
 - Modify the process design as appropriate
 - Incorporate results from bulk/residual drying technology studies to optimize the desiccant bed
 - Incorporate advanced CO₂ adsorbent materials as evaluations may indicate
- CO₂ Reduction
 - Incorporate and demonstrate post-processing stages to increase resource recovery
- Oxygen generation
 - Evaluate operational approaches and candidate replacements for a hydrogen sensor to simplify the hardware

Conclusion

- An ISS-derived AR subsystem architecture is feasible
 - Equivalent or better performance demonstrated
- Mass and volume reduction can be achieved
 - 12 kg and 15 liters by using distributed TCC components
 - Potential exists for CO₂ removal bed component size reduction
 - Simplified oxygen generator operational approaches may yield further mass and volume reduction
- Opportunity exists to demonstrate a higher degree of resource mass closure by incorporating CH₄ post-processing technologies
- Optically-based major constituent monitoring demonstrated steady, reliable performance.

Further Reading

- Perry, J.L., Carrasquillo, R.L., and Harris, D.W. (2006) Atmosphere Revitalization Technology Development for Crewed Space Exploration. 44th AIAA Aerospace Sciences Meeting and Exhibit. AIAA-2006-140. Reno, Nevada, January 2006.
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- Perry, J.L., Abney, M.B., Knox, J.C., Parrish, K.J., Roman, M.C., and Jan, D.L. (2012) Integrated Atmosphere Resource Recovery and Environmental Monitoring for Deep Space Exploration. AIAA 2012-3585, 42nd International Conference on Environmental Systems, San Diego, California.