



Functional Performance of an Enabling Atmosphere Revitalization Subsystem Architecture for Deep Space Exploration Missions

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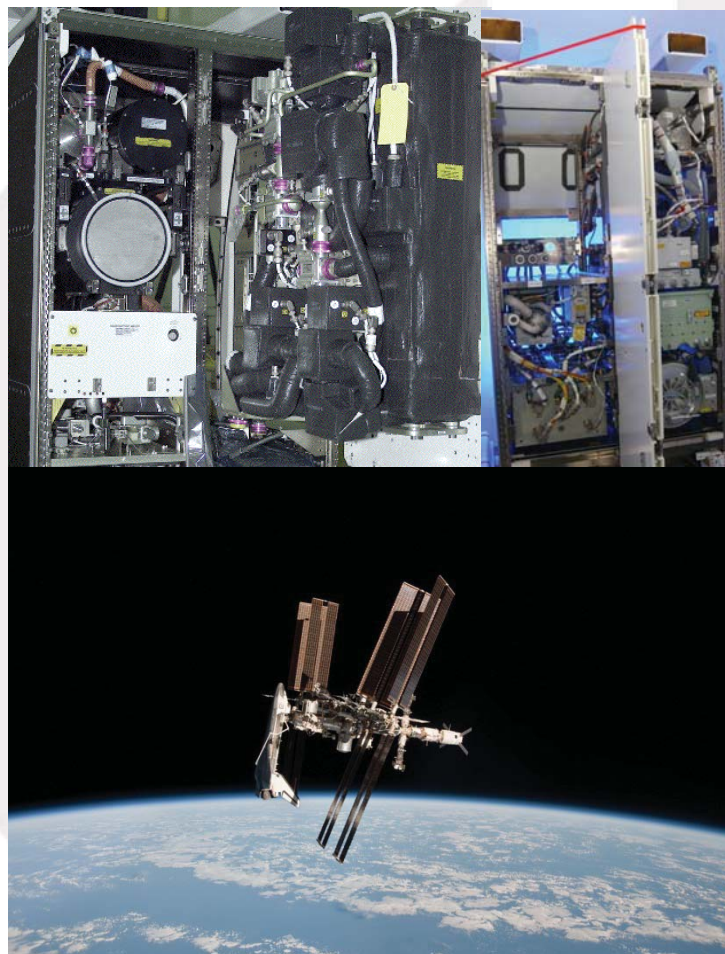
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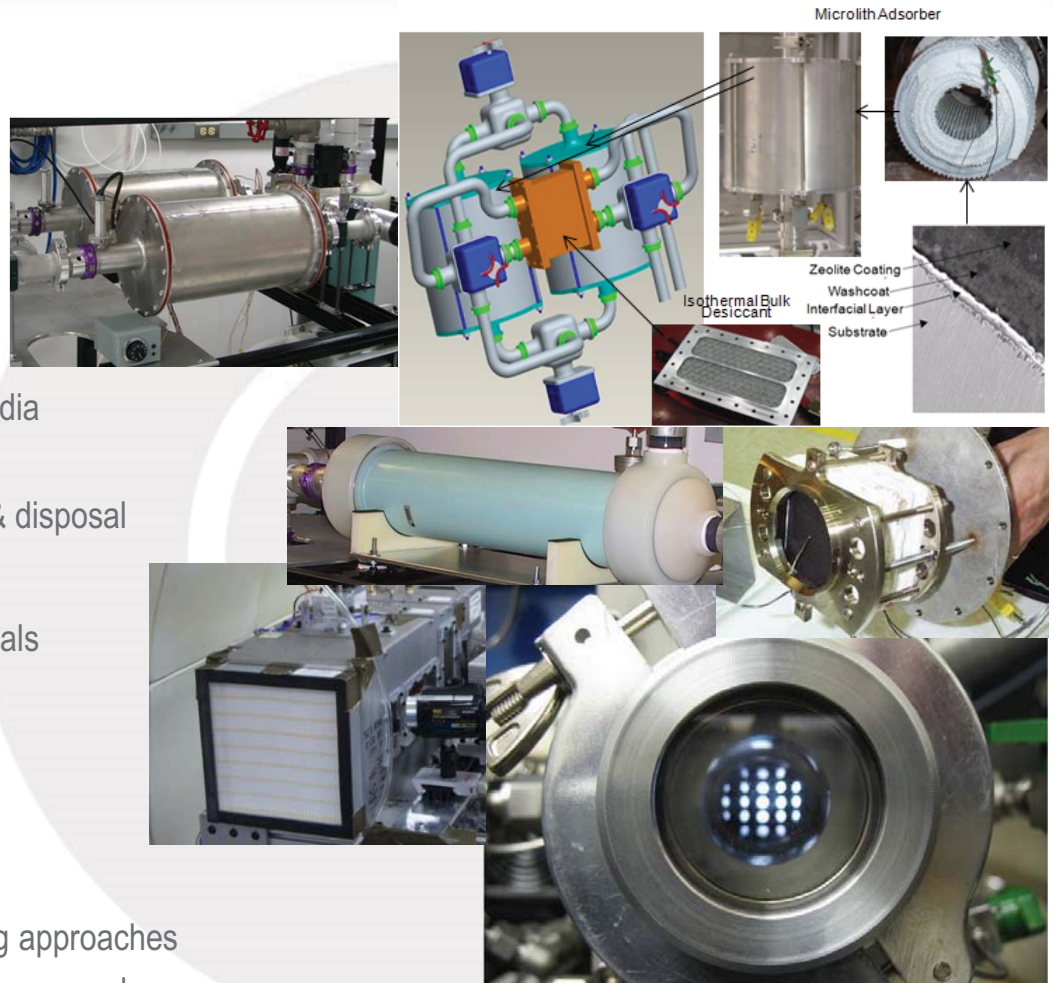
Technological Advancement Objectives

- To demonstrate an evolved ISS atmosphere revitalization (AR) subsystem architecture to enable deep space exploration
 - Improve reliability & maintainability
 - Reduce consumable mass
- To maximize commonality across missions and vehicles
- To mature process technologies for flight programs
 - Reduce technical risk and cost
- To develop modular resource recovery technologies



Strategic Improvements

- Cabin ventilation
 - Quiet fan design principles
- Carbon dioxide removal
 - Durable adsorbent media
 - Process air drying
- Trace contaminant control
 - Alternative high capacity adsorbent media
 - Structured oxidation catalysts
 - Low maintenance particulate filtration & disposal
- Oxygen supply
 - Long-lived electrolysis cell stack materials
 - Alternative process control approaches
- Oxygen recovery
 - Reduction byproduct processing
- Environmental monitoring
 - Alternative major constituent monitoring approaches
 - Alternative trace constituent monitoring approaches
 - Microbial & particulate monitoring techniques



Test Facility Overview

- **Characteristics**
 - 90.6 m³
 - Stainless steel
 - Vacuum-capable
- **Test support capabilities**
 - Metabolic simulation
 - Trace contaminant injection
 - Temperature and humidity control
 - Space vacuum resource simulation
 - Gas sample acquisition and analysis
 - Major constituent monitoring
 - Total pressure and atmosphere composition control
 - Process control & data acquisition/archiving
 - LabVIEW
 - Payloads and Components Real-time Automated Test System (PACRATS)
- **Analytical Instrumentation – VOC Analysis**
 - Agilent 5890 GC with a flame ionization detector coupled with a Markes TT24-7 Thermal Desorption System autosampler
 - Agilent 7890 GC with flame ionization and mass selective detectors coupled with a Gerstel Thermal Desorption System
 - MKS Multigas™ 2030 Fourier Transform Infrared Spectrometer
- **Analytical Instrumentation – Major Constituent Monitoring**
 - Oxigraf Model O2 analyzer – solid-state laser diode absorption
 - Sable Systems CA-2A analyzer – solid-state infrared absorption
 - Sable Systems RH-100 monitor – solid-state thin film capacitance



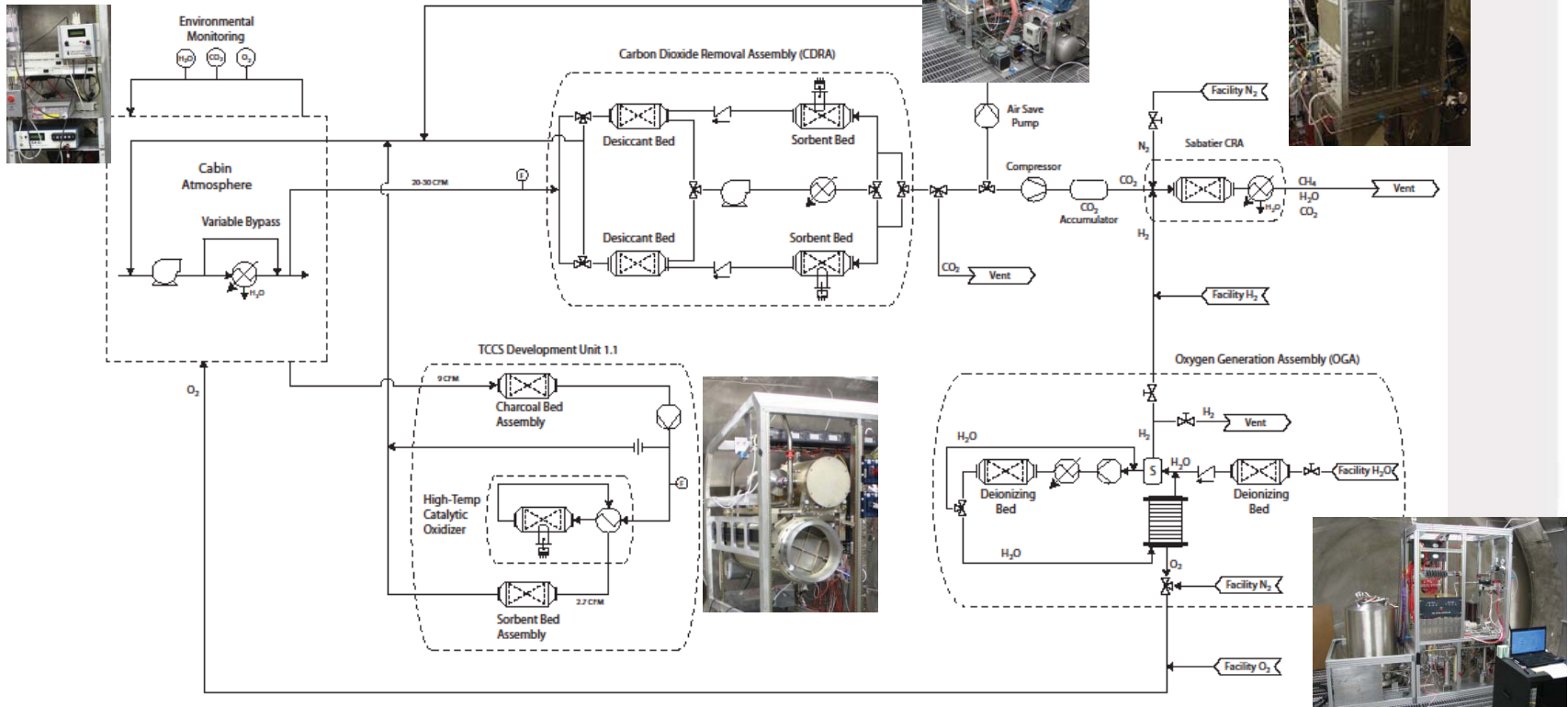
PARAMETER	RANGE
Total pressure	400-933 Pa gauge
Oxygen partial pressure	20.58±0.14 kPa
Carbon dioxide partial pressure	400±67 Pa
Temperature	21±2.8 °C
Relative humidity	50±5%
Contaminant injection rate	230 mg/hour*

*Percent by mass: methanol (10.7), ethanol (67.1), 2-propanol (4.8), ethanal (7.6), dimethylbenzene (2.3), dichloromethane (1.3), 2-propanone (6.2)




















ISS Architecture Testing Objectives

- Demonstrate functional performance of the basic ISS AR subsystem using the CDRA in CO₂ vent mode and the TCCS operating in parallel.
- Demonstrate the partial functional performance of the basic ISS AR subsystem when operating in a resource recovery mode that includes integration with CO₂ conditioning, storage, and reduction equipment.
- Investigate propagation of trace contaminants through the core ISS AR subsystem equipment with emphasis on the CDRA and CO₂ conditioning and storage equipment.
- Demonstrate the full resource recovery functional performance of the ISS AR subsystem including the CO₂ removal, CO₂ conditioning and storage, CO₂ reduction and post-processing, oxygen generation, and trace contaminant control functions.

ISS Architecture Testing



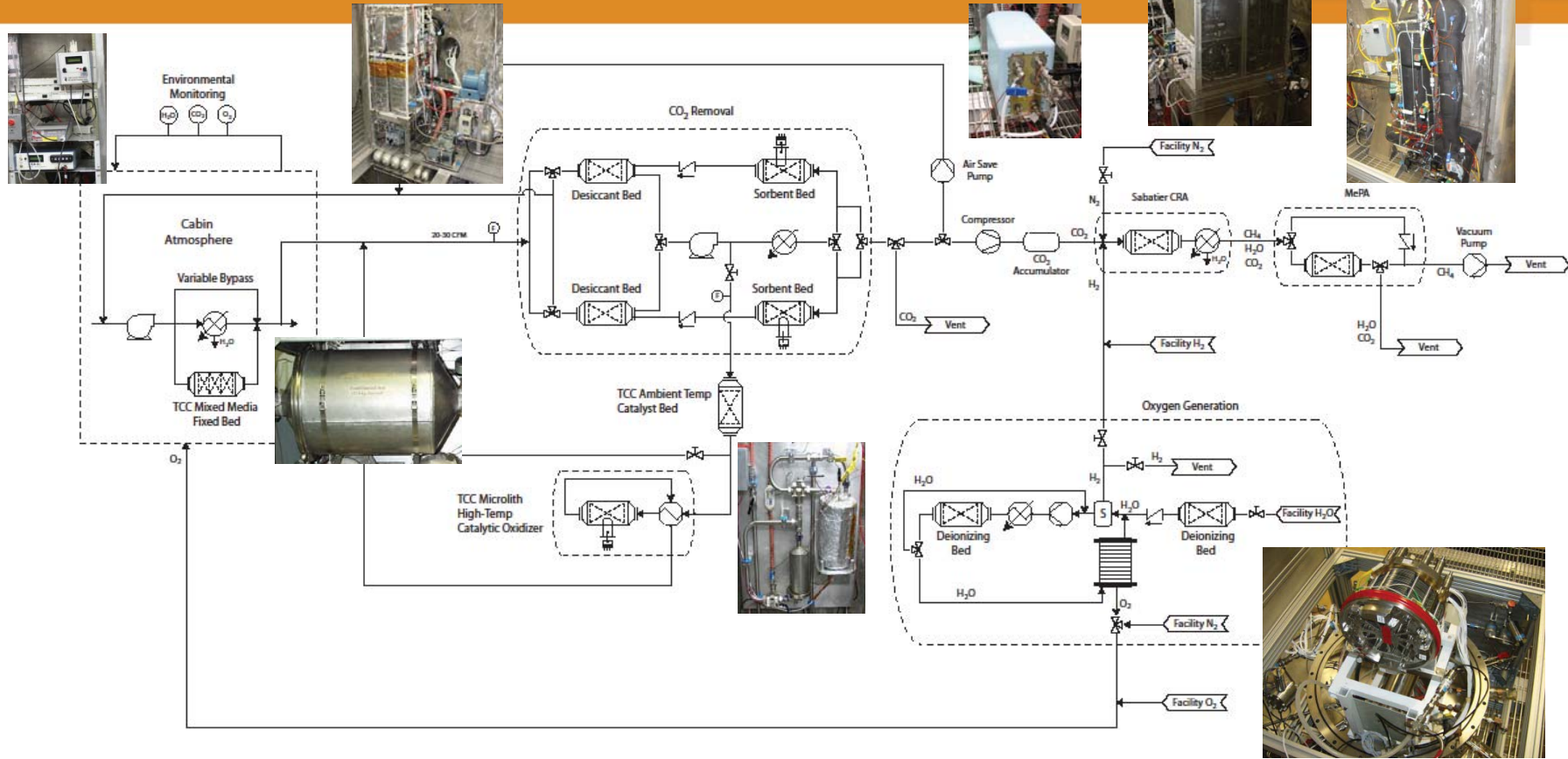
Symbols

	Packed bed		Check valve		Pump		Electrolysis Stack
	Heater		Three-way automatic control valve		Compressor		Accumulator
	Cooler		Two-way hand-operated valve		Blower		Separator
	Recuperative Heat exchanger		Dewpoint analyzer		Flowmeter		Orifice
	Condensing Heat exchanger		Carbon dioxide analyzer		Oxygen analyzer		




















Cycle 1 Testing Objectives

- Demonstrate integrated modified ISS subsystem operation
 - Oxygen generation
 - CO₂ removal
 - Trace contaminant control
 - Major constituent monitoring
 - CO₂ reduction with partial post-processing first stage to purify methane
- Demonstrate CO₂ management compressor operation
 - Observe how CO₂ removal process valve sequencing affects overall CO₂ reclamation efficiency for various modes of operation.
- Determine process fluid purities
 - Product CO₂ from the CDRA-4 sorbent beds.
 - Product oxygen and hydrogen from the OGA.
 - Product water from the Sabatier-based CRA.
- Determine the effect cabin atmosphere leakage and/or atmospheric major constituent inclusion on the CDRA CO₂ product may have on CRA performance.

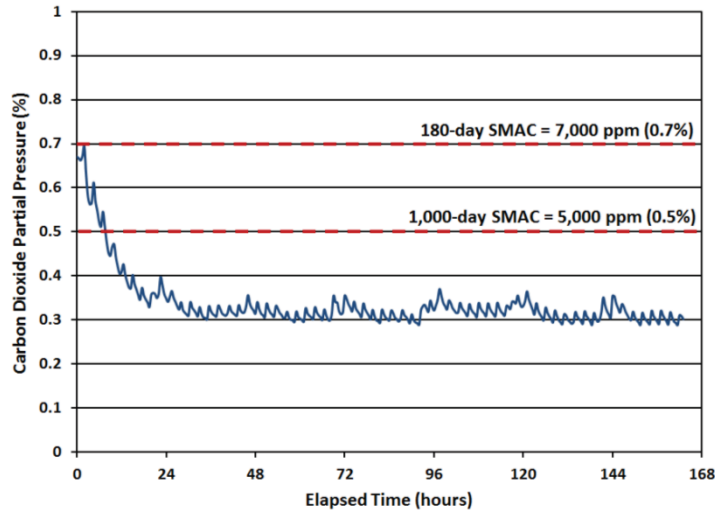
Cycle 1 Integrated Process Architecture



Symbols

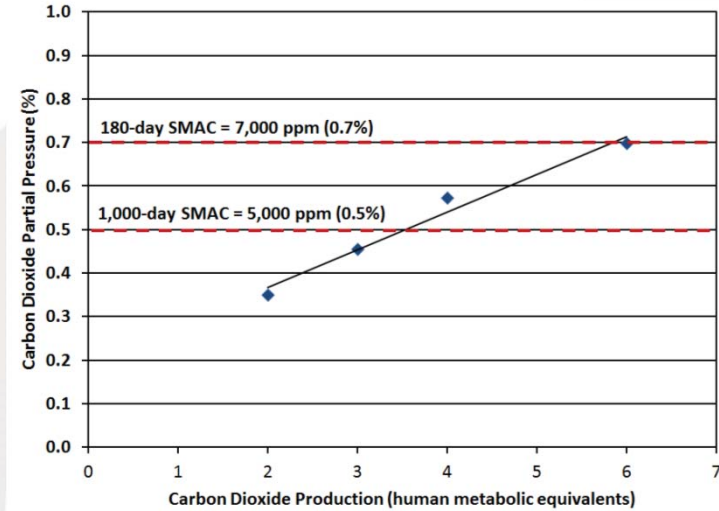
	Packed bed		Check valve		Pump		Electrolysis Stack
	Heater		Three-way automatic control valve		Compressor		Accumulator
	Cooler		Two-way hand-operated valve		Blower		Separator
	Recuperative Heat exchanger		Dewpoint analyzer		Flowmeter		Orifice
	Condensing Heat exchanger		Carbon dioxide analyzer		Oxygen analyzer		

Results: CO₂ Removal



ISS Architecture

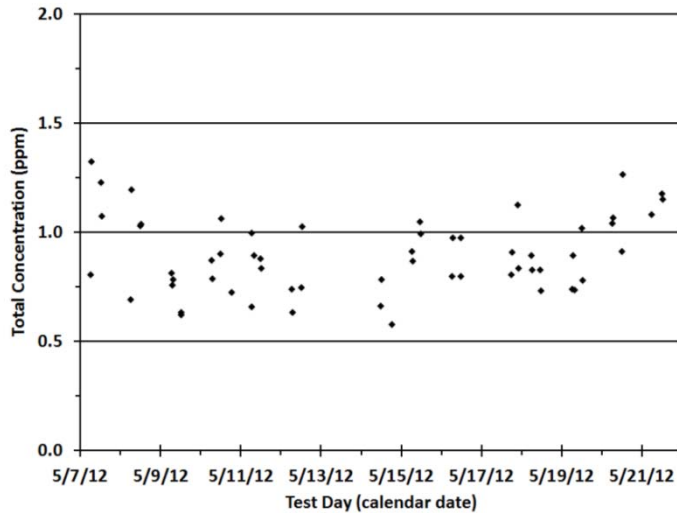
- ASRT zeolite 5A
- 34.6 m³/h flow
- 3-CM load
- 59% removal efficiency
- 0.34% CO₂ partial pressure



Cycle 1 Architecture

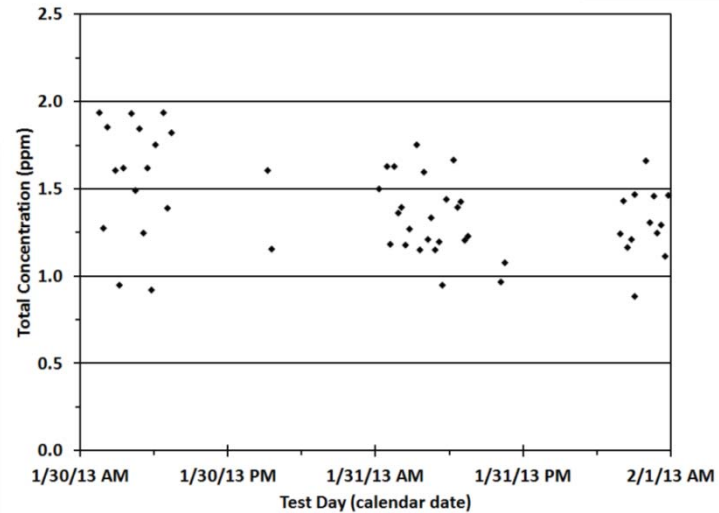
- RK-38 zeolite 5A
- 28.8 m³/hour flow
- 2-CM to 6-CM load
- 59% removal efficiency
- 0.45% CO₂ partial pressure

Results: Trace Contaminant Control



ISS Architecture

- Stand-alone TCC assembly
 - Carbon bed containing B-S Type 3032
 - Catalytic oxidizer containing Engelhard catalyst pellets
- 15.3 m³/h through carbon bed/4.6 m³/h through catalytic oxidizer
 - 88% mean single pass removal efficiency
- Indication of ethanol breakthrough of the carbon bed near end of testing
- 1.14 ppm mean total contaminant concentration

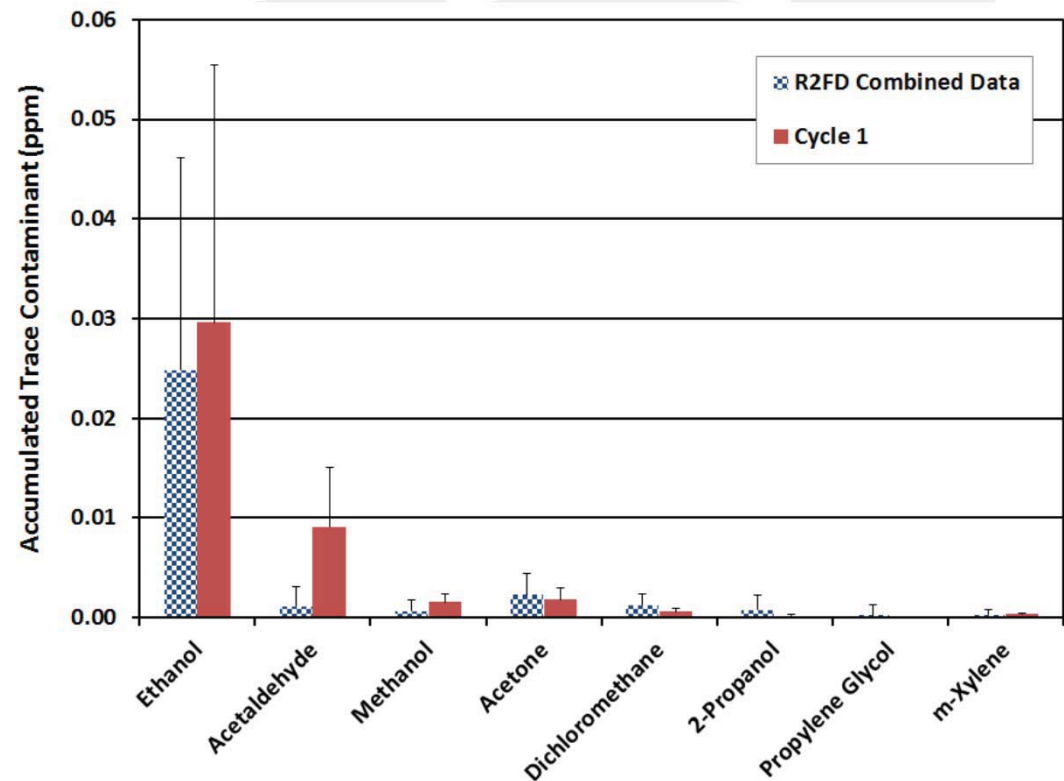


Cycle 1 Architecture

- Distributed TCC components
 - Carbon bed containing Chemsorb 1425 integrated with ventilation duct
 - Catalytic oxidizer containing Microlith™ catalytic reactor core integrated with CDRA
 - Eliminates electrical box and post-sorbent bed
- 8.5 m³/h through carbon bed
 - 91.2% mean single pass efficiency
- 5.9 m³/h through catalytic oxidizer
- 1.34 ppm mean total contaminant concentration

Results: CO₂ Reduction

- Water production rates consistent across architectures
 - ~2 ml/minute
- Consistent VOC loading in CO₂ feed
 - <0.03 ppm
- Suspected CO₂ reduction reaction poisons <<0.01 ppm
 - No indication of performance degradation during testing phases



Improvements for the Future

- Trace Contaminant Control
 - Full treatment of the ventilation flow with activated carbon cartridges mounted in main duct
- CO₂ Removal
 - Tune process parameters
 - Increase total flow by 6.6% to 37 m³/h to accommodate TCC catalytic oxidizer flow
 - Investigate half-cycle time and bed regeneration temperature modifications
 - Modify the process design as appropriate
 - Incorporate results from bulk/residual drying technology studies to optimize the desiccant bed
 - Incorporate advanced CO₂ adsorbent materials as evaluations may indicate
- CO₂ Reduction
 - Incorporate and demonstrate post-processing stages to increase resource recovery
- Oxygen generation
 - Evaluate operational approaches and candidate replacements for a hydrogen sensor to simplify the hardware

Conclusion

- An ISS-derived AR subsystem architecture is feasible
 - Equivalent or better performance demonstrated
- Mass and volume reduction can be achieved
 - 12 kg and 15 liters by using distributed TCC components
 - Potential exists for CO₂ removal bed component size reduction
 - Simplified oxygen generator operational approaches may yield further mass and volume reduction
- Opportunity exists to demonstrate a higher degree of resource mass closure by incorporating CH₄ post-processing technologies
- Optically-based major constituent monitoring demonstrated steady, reliable performance.

Further Reading

- Perry, J.L., Carrasquillo, R.L., and Harris, D.W. (2006) Atmosphere Revitalization Technology Development for Crewed Space Exploration. 44th AIAA Aerospace Sciences Meeting and Exhibit. AIAA-2006-140. Reno, Nevada, January 2006.
- Perry, J.L. (2007) Atmosphere Revitalization--Process Technology Maturation for NASA's Constellation Projects. Space Technology and Applications International Forum (STAIF 2007), Albuquerque, New Mexico, February 2007.
- Perry, J.L. and Howard, D.F. (2007) Spacecraft Life Support System Process Technology Maturation using Stage Gate Methodology. 37th International Conference on Environmental Systems. SAE 2007-01-3045.
- Perry, J.L., Bagdigian, R.M., and Carrasquillo, R.L. (2010) Trade Spaces in Crewed Spacecraft Atmosphere Revitalization System Development. AIAA 2010-6061, 40th International Conference on Environmental Systems, Barcelona, Spain.
- Perry, J.L., Abney, M.B., Knox, J.C., Parrish, K.J., Roman, M.C., and Jan, D.L. (2012) Integrated Atmosphere Resource Recovery and Environmental Monitoring for Deep Space Exploration. AIAA 2012-3585, 42nd International Conference on Environmental Systems, San Diego, California.