

DebriSat Project Update and Planning

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DebriSat Reporting Topics



- DebriSat Fragment Analysis Calendar
- Near-term Fragment Extraction Strategy
- Fragment Characterization and Database
- HVI Considerations
- Requirements Document

DebriSat Fragment Analysis Calendar



- The panel X-ray task shown in the table below should be completed by the end of FY16 (09/30/16).
- Fragment extraction and characterization will continue will continue throughout the duration of the project.
- DebrisLV activities are of lower priority, and may begin in FY17.

Task	FY 15			FY16			FY17				FY18					
Task	Q1	Q2	Q3	Q4	Q1	Q2	Q3	Q4	Q1	Q2	Q3	Q4	Q1	Q2	Q3	Q4
X-ray: DebriSat																
Extraction: DebriSat				-		:	 	:								
Characterization: DebriSat																
X-ray: DebrisLV											→					
Extraction: DebrisLV											-		,			
Characterization: DebrisLV														(

Near-term Fragment Extraction Strategy





- Pre-test foam panel placement is displayed in this DebriSat before impact photo
- Low-density polyurethane foam panels , each < 2" thick were installed inside target chamber one on top of the other (the most dense panel on the bottom)
- Panel colors indicate position in the chamber only
- Wide panels were placed horizontally and vertically
- Narrow panels were placed at 45deg to allow nearly full coverage of the interior of the chamber
- A 30deg debris cone spread was expected by the operators to the end of the chamber
- The next slide offers a cartoon push-out view of the panel placement with a progress report of panel preparation



UF Foam Panel Processing Plan UF as of 2/4/16



Panels are stacked sideways in figure

Panel layers 0 (loose bagged material if any) and 1 (least dense on top) to 8 (densest panel on bottom)

+ is the pre-test position of DebriSat in impact chamber

Row 3 is chosen as the first chamber row, "ring" to be analyzed

Fragment Characterization and Database



- The DebriSat project group took care to build a spacecraft with components that mimic modern spacecraft.
- Plan is to extract \geq 2 mm fragments from foam panels/loose pieces/dust
- Plan is to recover at least 90% of the total DebriSat mass from the fragments
- Debris collected so far ~112K, Debris recorded so far >107K
- The current estimate for the <a>2mm DebriSat fragments to be collected is more than 200,000
 - > Assign each individual fragment a unique identification number
 - A label of 2-D or 3-D is given to each fragment based on the means of analysis of fragment dimensions that will be performed
 - > Fragment physical parameters filling out the database include material/color, shape
 - Fragment derived parameters include characteristic length, volume, average cross sectional area, mass, bulk density
- The small fragment population so far favors tens of thousands of CFRP (Carbon Fiber Reinforced Polymer) fragments created from the splintering DebriSat spacecraft skin and interior inserts. These fragments favor the 2-D imaging techniques



Sample 2-D and 3-D Small Fragments





- Fragments labeled

 2, and 3 are
 CFRP splinters.
 These will be
 defined as 2-D
 objects.
- Wire fragments 4, 5 and the wire bundle (8) will likely treated as 3-D.
- The nugget (6) and charred bent plate (7) are 3-D.
- The next slide displays the 2-D and 3-D apparatuses

2-D vs 3-D



2-D Imaging



3-D Reconstruction



UF Database



- Investigating whether "needle-like fragment" dimensions (i.e., thickness is assumed to be negligible) can be approximated with simple 2-D assumption
- Further research is required to determine what the average thickness of the CFRP fragments are, to provide a range of expected thicknesses

UF Database (Homepage)





UF database plans to include single-tagged fragments ≥ 2mm in maximum dimension, original locations within panels, material, shape, mass, volume, CS area, characteristic length, and multiple photos

UF Database (ID, Locations, Measurements)



University of Florida - DCS ×		- D
→ C ní D 10.244.15.144/dcs/secured/add_debris.php		会 🔅
Debris Categorization System	JACOBS	8
Add Debris Logged in as: joeykleespies		Home
IDENTIFICATION		Debris
Project: Debitsat V Box #:		Foam
LOCATION		Learnt
Related Foam Panel:	LIPRNG Row 1 Row 2	Logout
-MEASUREMENTS	Row 3 Row 4	
Material: • Shape: •	Row 5 DNRNG	
Mass: g Volume: mm ³	Density: gimm ³	
X _{DDM} : mm Y _{DDM} : mm	Z _{DIM} : man	
L _{CHAR} : mm CS Area: mm ²		
-MISCELLANEOUS Foam Attached:		

HVI Considerations



- Fragments in the UF Database are categorized by specific subjective criteria, for example, material, shape, size, color, location in panels) for the purpose of estimating lethality of these fragments in on-orbit collisions
 - In many cases their respective origins can not be determined with certainty.
 - CFRP highlight in the end. Invisible to radar....

Shape
flat plate
bent plate
straight/needle/rod/cylinder
bent/needle/rod/cylinder
parallelepiped/nugget/spheroi d
flexible /MLI

Color
silver
red
black
gold
light blue
royal blue
magenta
clear
burnt/charred

Material
aluminum
stainless Steel
glass
printed circuit board
copper
MLI/kapton
carbon fiber
plastic
multi-material

National Aeronautics and Space Administration

HVI Considerations



Debris Detection Software and Fragment Extraction





CAD model of Debrisat without MLI Test Plan 2014-03-13

