# **INSTABILITY** ANALYSIS OF CYLINDRICAL **SHELLS**

## **UNDER** HYDROSTATIC PRESSURE

### By George Herrmann

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#### **SUMMARY**

**To determine** the elastic **buckling pressure** of simply supported various versions of linear bending theories have been employed in the past. For certain shell dimensions, however, the expressions commonly used may yield substantially differing results. In what follows, recent work on this problem by A. E. Armenakas and the writer is briefly reviewed. This work consisted primarily in employing a general bending theory of circular cylindrical shells under the influence of initial stress, developed earlier by the same authors, to re-examine the problem mentioned, and compare the results with those of previous investigations. The outcome was the establishment of a simple but **investigations** of the buckling pressure applicable to a window accurate expression for the buckling **pressure and all dimensions.** range **of** shell **dimensions.**

In the past, various versions of linear bending theories have<br>been employed in establishing the buckling value of the external pressure acting on circular cylindrical shells. One of the earliest investigations by von Mises (ref. 1) resulted in a simple expression for the critical uniform lateral pressure (no axial stress) that has been utilized extensively. Timoshenko (ref. 2) indicated that this formula is in close agreement with a more intricate formula obtained by Flügge (ref. 3). Much later, von Mises also considered the case of **a** shell under all around pressure (axial stress equals one half of the a shell are shell are (and mesented a formula for the critical  $\frac{1}{2}$  stress (see  $\frac{1}{2}$  page  $\frac{1}{2}$ ).

More recently, Batdorf (ref. 4) reinvestigated this **problem** and von Mises for the case of all around pressure, and a somewhat different expression for the case of only lateral pressure. As is well known, expression for the challs whose dimensions conduce to buckling in mode however, for shells whose **dimensions** conduce to buckling in modes with

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 $\frac{1}{2}$  pressure (see ref. 2).

a small number of circumferential waves, the assumptions made in deriving the Donnell equations are not valid and, concomitantly, for such shells the Batdorf results are inaccurate. Mushtari and Galimov (ref. 6) present an analysis of the same problem also using the Donnell equations and their results are thus subject to the same limitations.<br>Loo (ref. 7) attempted to use, in establishing the buckling pressure, an extended Donnell equation which included the effect of transverse shear forces on the equilibrium in the circumferential direction. Calculations based on this modification appear, for long shells, to be in reasonable agreement with the results obtained by von Mises.

(ref. 6) **present** an analysis **of** the ssne problem also **using** the Donnell exted to the substantial formulas have been referred to frequently in the literature. For certain ranges of shell dimensions, however, they yield substantially differing results. Thus, it is and the effect substantially differing results. Thus, it is shear that it is some the examination of the cuckling of circuls cylindrical shells under hydrostatic pressure was desirable. For this purpose, the bending theory of cylindrical shells, under the influence of a general state of initial stress, presented recently by Herrmann and Armenakas (ref. 8), was used to establish the value of the critical pressure acting on a cylindrical shell. The results were compared with those of previous investigations and a simplified but accurate expression for the buckling pressure, applicable to a wide range of shell everious the systematic resolution of a systematic reduction of the buckling of shell count independent of the simple supports. A detailed account of this work, including considerations concerning the character of the **external pressure, is given in ref. 9.** 

#### pressure **acting on** a cylindrical shell. The results ve\_e compared with those **of** previous investigations and a simplified but **accurate** ex-

**end** Armenakas (re\_. 8), was used to establish the value **of the** critical

- L  $shell1$  length **of** this work, including considerations concerning the character **of** the
- $\mathbf R$ shell radius
- h shell thickness
- $E$ Young's modulus
- Poisson's ratio N
- D
- number of circumferential waves (lobes)  $\mathbf n$
- number of axial half-waves m



Subscript:

cr critical

## CYI\_CAL SHELLS UNDER **HYDROSTATIC** PRESSURE

## Exact Expression for Buckling Pressure

Hydrostatic pressure (as **distinct,** for example, from constantdicular to the area element and having during the process of deformation a constant magnitude per unit actual area. It can act on the ation a constant magnetic constant of the actual area. N or at the end lateral surface of the shell, inducing a stress N, **the** shell, inducing a stress T.

 $\bar{N}$   $=$   $\bar{2}$  corresponds to all around press

 $\frac{T}{N} = 0$  to lateral pressure only.<br>If the shell is assumed to be simply supported, i. e. the circumferential displacement, the radial displacement, the axial force and the axial bending moment are all zero, then the following expression the axial behavior are all  $\alpha$  is obtained, on the be for the critical **hydrostatic** pressure Pcr is **obtained,** on the basis of

the equations presented in ref. 8.

$$
P_{cr} = K \frac{\pi^2 D}{L^2 R}
$$

$$
G = n^{4}(\beta^{2} + m^{2})^{4} + 12Z^{2}n^{4}m^{2}\pi^{-4} + \beta^{8}(1 - 2n^{2})
$$
  
+  $2m^{2}\beta^{6} [2(1 - 2n^{2}) + v(n^{2} - 1)] - 6m^{4}n^{2}\beta^{4} - 2m^{6}n^{2}\beta^{2}v$   

$$
H = \beta^{2}n^{4}(\beta^{2} + m^{2}) + \frac{T}{N}(1 + s)m^{2}n^{4}(\beta^{2} + m^{2})^{2} + \frac{T}{N}(1 + s)m^{2}n^{2}\beta^{2} [ \beta^{2} + m^{2}(3 + 2v)] - \beta^{4}n^{2}(\beta^{2} + 3m^{2}) + m^{2}n^{2}s [ \beta^{4} + n^{2}(\beta^{2} + m^{2})]
$$

 $K = \frac{G}{H}$ 

Terms involving s result from taking into account the fact that the pressure is not acting on the middle surface of the shell. The above  $T_{\text{max}}$  involving s result from taking into account the fact that **py** riugge (rei. )). For thin shells this expression is unnecessarily complicated and may be simplified considerably.

### Approximate expression for Buckling Pressure

It is well known that shells having a small curvature parameter Z buckle into a large number of circumferential waves, and therefore may be disregarded as compared to unity. The buckling coefficient buckle into a large **n\_mber of** clrcumferential **waves,** and therefore  $\mathbf{v}$ .

$$
K = \frac{(1+\beta^2)^2}{\frac{T}{N}+\beta^2} + \frac{12Z^2}{\pi^4(1+\beta^2)^2} \frac{(\frac{T}{N}+\beta^2)}{(\frac{T}{N}+\beta^2)}
$$

For  $\frac{T}{N} = \frac{1}{2}$ , this expression is identical (ref. 2, p. 498) and Batdorf (ref. 4). For the range of shell dimensions for which the above equation is not valid, that is for shells buckling into a small number of circumferential waves  $n$ ,  $\beta^2$  is large as compared to unity. Therefore, the exact expression for K may be simplified by disregarding in it terms not included in the approximate expression above, whose order of magnitude is  $\frac{2}{7}$  times that of term simplified by disregarding in it terms **mot** included in the approximate retained. The following expression for the buckling coefficient is then obtained, valid for a wide range of shell dimensions

retained. The following **expression** for the buckling coefficient is

$$
K = \frac{(1+\beta^2)^2 (n^2-1)}{(\frac{n}{N}+\beta^2)n^2} + \frac{122^2n^2}{\pi^4(1+\beta^2)^2 (\frac{n}{N}+\beta^2) (n^2-1)}
$$

**In order to** evaluate this **buckling** coefficient for a ahell **of** specified dimensions, it is necessary to establish first the mailland mode (mmmper of circumferential waves n) which licence and smallest value of K for given  $\frac{1}{N}$ ,  $\beta$  and Z. This can be done by repeated calculations, **or by** using the chart given in ref. **9,** which is an extension of curves **presented** in ref. I0.

#### Concluding Remarks

An evaluation of earlier work is presented in ref. 9 and the conclusion was reached that neither von Mises' nor Loo's expressions conclusion was reached that neither you history of that the state of the state are **based** on consistently admissible shell theories, sadden, sad the shell theories, sadden, sad that the shell the shell theories, sadden, yield, for certain shell **dimensions,** incorrect results. Dom\_ell-type equations, such as used **by Batdorf** (ref. 4) cannot **be** expected to yield good results for **long** shells.

**F\_the\_,** it is **of** interest **to point out that the new** expression for **the buckling pressure** cannot **be derived directly** fr\_ **a** consistent simplified shell theory. It appears that this expression can be ob-<br>tained only on the basis of the complete bending theory, followed by a **tained** only on the basis of the complete building coefficient. In simplification of the expression for the **buckling** coefficient. future work it might **be** worthwhile to attempt construotiong **a** similar expression for **the** buckling coefficient valid for more **realistic boundary** conditions.

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