

UNCLASSIFIED INFORMATION

ON THE DESTABILIZING EFFECT OF DAMPING
IN NONCONSERVATIVE ELASTIC SYSTEMS¹

by

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GPO PRICE \$ _____
CSFTI PRICE(S) \$ _____
Hard copy (HC) 200
Microfiche (MF) 500

RECEIVED
OCT 13 11 34 AM '64
AERONAUTICAL ENGINEERING

FACILITY FORM 808

(ACCESSION NUMBER)	N65 32152
(PAGES)	1
(CATEGORY)	32
(NASA CR OR TMX OR AD NUMBER)	WS93576

¹ This research was supported by the National Aeronautics and Space Administration under Grant NsG 605.

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SUMMARY

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The destabilizing effect of linear viscous damping in a non-conservative elastic system is investigated by studying the roots of the characteristic equation in addition to the stability criteria and by introducing the concept of degree of instability. A generic relationship between critical loadings for no damping and for slight damping as well as vanishing damping is established. It is found that while the presence of small damping may have a destabilizing effect, proper interpretation of the limiting process of vanishing damping leads to the same critical load as for no damping.

Author

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MAY 1964

Introduction

It has been discovered by Ziegler [1] a little more than a decade ago, that internal damping may have a destabilizing effect in nonconservative elastic system. He considered a double pendulum with viscoelastic hinges as a model of an elastic bar with internal damping and let a tangential force act at the free end. The critical loading obtained in complete absence of damping was found to be considerably higher than by including damping at the outset of the analysis and then letting the damping coefficients approach zero (vanishing damping) in the expression for the critical force.

This rather surprising and paradoxical finding was ascribed in later studies by Ziegler [2], [3] to the possibility that internal damping is inadequately represented by linear damping forces which are linear combinations of the generalized velocities and that the hysteresis effect should be taken into account.

The destabilizing effect of damping was further elaborated upon by Bolotin [4] who considered a general two-degree-of-freedom system not related to any particular mechanical model and who found additionally that the destabilizing effect in the presence of slight and vanishing damping is highly dependent on the relative magnitude of damping coefficients in the two degrees freedom.

It is the aim of the present investigation to make an attempt at supplying some additional insight into the destabilizing effects of linear velocity-dependent damping in nonconservative systems, without raising the question here as to the suitability of this damping mechanism

for a realistic system. For this purpose the system discussed by Ziegler is reconsidered, and not only the stability conditions are investigated but also the roots of the characteristic equations themselves. Plots of these roots for various ranges of loading illustrate graphically how the paradoxical effects of vanishing damping are generated. Further, the results of the mathematical stability investigations are interpreted in physical terms by introducing the concept of degree of instability.

These concepts permit to carry out a gradual transition from the case of small damping to the case of vanishing damping and relate them to the case of no damping. Finally, some remarks are made with regard to possible behavior of an elastic bar with distributed parameters.

The Model

We consider a double pendulum, Fig. 1, composed of two rigid weightless bars of equal length l , which carry concentrated masses $m_1 = 2m$, $m_2 = m$. The generalized coordinates φ_1 , φ_2 are taken to be small. A load P applied at the free end is assumed to be acting at an angle φ_2 (follower force). At the hinges the restoring moments $c\varphi_1 + b_1\dot{\varphi}_1$ and $c(\varphi_2 - \varphi_1) + b_2(\dot{\varphi}_2 - \dot{\varphi}_1)$ are induced.

The kinetic energy T , the dissipation function D , the potential energy V , and the generalized forces Q_1 and Q_2 are:

$$T = \frac{1}{2} m l^2 (3\dot{\varphi}_1^2 + 2\dot{\varphi}_1\dot{\varphi}_2 + \dot{\varphi}_2^2)$$

$$D = \frac{1}{2} b_1 \dot{\varphi}_1^2 + \frac{1}{2} b_2 (\dot{\varphi}_1^2 - 2\dot{\varphi}_1\dot{\varphi}_2 + \dot{\varphi}_2^2)$$

$$V = \frac{1}{2} c (2\varphi_1^2 - 2\varphi_1\varphi_2 + \varphi_2^2)$$

$$Q_1 = Pl (\varphi_1 - \varphi_2)$$

$$Q_2 = 0$$

Lagrange's equations in the form

$$\frac{d}{dt} \left(\frac{\partial T}{\partial \dot{\varphi}_i} \right) + \frac{\partial D}{\partial \dot{\varphi}_i} - \frac{\partial T}{\partial \varphi_i} + \frac{\partial V}{\partial \varphi_i} = Q_i \quad (i = 1, 2)$$

are employed to establish the linear equations of motion

$$3ml^2\ddot{\varphi}_1 + (b_1 + b_2)\dot{\varphi}_1 - (Pl - 2c)\varphi_1 + ml^2\ddot{\varphi}_2 - b_2\dot{\varphi}_2 + (Pl - c)\varphi_2 = 0$$

$$ml^2\ddot{\varphi}_1 - b_2\dot{\varphi}_1 - c\varphi_1 + ml^2\ddot{\varphi}_2 + b_2\dot{\varphi}_2 + c\varphi_2 = 0$$

which, upon stipulating solutions of the form

$$\varphi_i = A_i e^{i\omega t} \quad (i = 1, 2)$$

yield the characteristic equation

$$p_0 \omega^4 + p_1 \omega^3 + p_2 \omega^2 + p_3 \omega + p_4 = 0$$

with the coefficients

$$p_0 = 2$$

$$p_1 = B_1 + 6B_2$$

$$p_2 = 7 - 2F + B_1 B_2$$

$$p_3 = B_1 + B_2$$

$$p_4 = 1$$

and the dimensionless quantities

$$\Omega = l(\frac{m}{c})^{\frac{1}{2}} \omega$$

$$B_i = \frac{b_i}{l(\text{cm})^{\frac{1}{2}}} \quad (i = 1, 2)$$

$$F = \frac{Pl}{c}$$

In the absence of damping ($B_1 = B_2 = 0$), the characteristic equation is a biquadratic

$$2\Omega^4 + (7-2F)\Omega^2 + 1 = 0$$

Critical Loads

From the assumed form of the time-dependence for the coordinates φ_i and on the basis of the kinetic stability criterion, it is evident that if all four roots of the characteristic equation are distinct, the necessary and sufficient conditions for stability are that the real roots and the real parts of the complex roots should be all negative or zero. In case of equal roots the general solution of φ_i will have terms which contain powers of t as a factor. If the real parts of equal roots are negative, the system will be stable (vibration with decreasing amplitude), but if these real parts are zero or positive, stability will not exist (vibration with increasing amplitude).

Turning our attention first to the case of an initially undamped system, the four roots of the biquadratic equation as a function of F are

$$\Omega_{1,2,3,4} = \frac{1}{2} \left[\pm \sqrt{F - (7/2 - \sqrt{2})} \pm \sqrt{F - (7/2 + \sqrt{2})} \right]$$

and are plotted in Fig. 2. The projections of the root curves on the

real plane ($\text{Im}\Omega = 0$), the imaginary plane ($\text{Re}\Omega = 0$) and the complex plane ($F = 0$) are also shown in Fig. 2.

It is found that there will always be two roots with positive real part if $F > \frac{7}{2} - \sqrt{2} = 2.086 = F_e$. For $F = F_e$ there exist two pairs of equal roots whose real parts are all zero. Thus the system is unstable for $F \geq F_e$. For $F < F_e$ all roots are distinct and pure imaginary and thus the system is stable. A further discussion of an initially undamped system is presented in [5].

We consider next a slightly damped system, assuming $B_1 = B_2 = 0.01$. No simple expressions for the four roots of the quartic equation exist; the numerical results obtained are illustrated in Fig. 3 where a perspective view is supplemented by three projections on the same three planes as in Fig. 2. Two roots will have a positive real part for $F > 1.464 = F_d$.

Stability can be investigated directly without determining the roots of the characteristic equation by applying the Routh-Hurwitz criteria [6], which require that all coefficients p_j ($j = 0, \dots, 4$) of the characteristic equation and the quantities

$$S = p_1 p_2 - p_0 p_3$$

$$X = p_1 p_2 p_3 - p_0 p_3^2 - p_1^2 p_4$$

be positive. For positive damping these stability conditions are satisfied, provided

$$p_2 = 2[-F + \frac{1}{2}(7 + B_1 B_2)] > 0$$

$$S = 2(B_1 + 6B_2) \left\{ -F + \left[\frac{5(B_1 + 8B_2)}{2(B_1 + 6B_2)} + \frac{1}{2} B_1 B_2 \right] \right\} > 0$$

$$X = 2(B_1^2 + 7B_1 B_2 + 6B_2^2) \left\{ -F + \left[\frac{4B_1^2 + 33B_1 B_2 + 4B_2^2}{2(B_1^2 + 7B_1 B_2 + 6B_2^2)} + \frac{1}{2} B_1 B_2 \right] \right\} > 0$$

For the system to be stable F must satisfy the following three inequalities, where $\beta = B_1/B_2$, $0 \leq \beta \leq \infty$

$$F < \frac{7}{2} + \frac{1}{2} B_1 B_2$$

$$F < \frac{5(\beta+8)}{2(\beta+6)} + \frac{1}{2} B_1 B_2$$

$$F < \frac{4\beta^2 + 33\beta + 4}{2(\beta^2 + 7\beta + 6)} + \frac{1}{2} B_1 B_2$$

Since

$$\frac{5(\beta+8)}{2(\beta+6)} = \frac{5}{2} + \frac{5}{\beta+6} \leq \frac{10}{3} < \frac{7}{2}$$

$$\frac{4\beta^2 + 33\beta + 4}{2(\beta^2 + 7\beta + 6)} = \frac{5(\beta+8)}{2(\beta+6)} - \frac{\beta+3}{2(\beta+1)} < \frac{5(\beta+8)}{2(\beta+6)} < \frac{7}{2}$$

for whatever β in its range, it is evident that the critical load will be governed by the third inequality, i.e.,

$$\bar{F}_d = \frac{4\beta^2 + 33\beta + 4}{2(\beta^2 + 7\beta + 6)} + \frac{1}{2} B_1 B_2$$

which depends on the ratio as well as the magnitudes of the damping coefficients.

For $B_1 \ll 1$, as well as in the limit of vanishing damping, \bar{F}_d becomes

$$F_d = \frac{4\beta^2 + 33\beta + 4}{2(\beta^2 + 7\beta + 6)}$$

which is highly dependent on β and is in general smaller but never larger than F_e . The ratio of F_d to F_e versus β is plotted in Fig. 4. It is noted that when $\beta = 4 + 5\sqrt{2} = 11.07$, F_d/F_e reaches its maximum value 1. The destabilizing effect is thus eliminated in this particular case, similar to that found by Bolotin [4]. For $\beta = 0$, F_d/F_e reaches its minimum value 0.16; i.e., the maximum destabilizing effect is about 84% in the present two-degree-of-freedom system.

The Case of Vanishing Damping

The two disparate values of the critical load for no damping ($B_1 = 0$) and vanishing damping ($B_1 \rightarrow 0$) justify a more detailed investigation of the limiting process as the damping coefficients approach zero.

Let us examine first the limiting process for the roots of the characteristic equation. It can be shown with the aid of the theory of equations [7] that if $B_1 \ll 1$ and $F < 4.914$ this equation will have four complex roots. Let these roots be

$$\Omega = \begin{cases} \gamma_1 \pm i\gamma_2 \\ \lambda_1 \pm i\lambda_2 \end{cases}$$

Then one can write [6], [7]

$$2(\gamma_1 + \lambda_1) = -\frac{p_1}{p_0}$$

$$4\gamma_1\lambda_1 [(\gamma_1+\lambda_1)^2 + (\gamma_2+\lambda_2)^2][(\gamma_1+\lambda_1)^2 + (\gamma_2-\lambda_2)^2] = \frac{X}{p_0^3}$$

where p_0 , p_1 and X are as defined earlier. For vanishing damping

$$\begin{cases} \gamma_1 + \lambda_1 = 0 \\ \gamma_1\lambda_1 [(\gamma_1+\lambda_1)^2 + (\gamma_2+\lambda_2)^2][(\gamma_1+\lambda_1)^2 + (\gamma_2-\lambda_2)^2] = 0 \end{cases}$$

Hence $\gamma_1 = -\lambda_1$, $\gamma_2 = \lambda_2$

or $\gamma_1 = \lambda_1 = 0$

Thus

$$\Omega = \begin{cases} \pm i\gamma_2 \\ \pm i\lambda_2 \end{cases} \quad \text{or} \quad \Omega = \begin{cases} \gamma_1 \pm i\gamma_2 \\ -\gamma_1 \pm i\gamma_2 \end{cases}$$

and a substitution of these four roots into the characteristic equation will show that they are the same as in the case of no damping.

In the case of $F \geq 4.914$, the four roots will be all real for small B_1 . Let

$$\Omega = \begin{cases} u_1 \pm u_2 \\ v_1 \pm v_2 \end{cases}$$

In the limit of vanishing damping one can show similarly that either $u_1 = v_1 = 0$ or $u_1 = -v_1$ and $u_2 = v_2$. For either alternative, substitution into the characteristic equation reveals that the roots are the same as in the case of no damping.

Thus the conclusion is reached that whatever F the roots of the characteristic equation for no initial damping ($B_1 = 0$) are identical to those of vanishing damping ($B_1 \rightarrow 0$). This implies that the motions

of the system, for some given initial conditions, and whatever F , will be identical in the case of no damping ($B_1 = 0$) and vanishing damping ($B_1 \rightarrow 0$).

We focus attention next on the loading F in the two cases and before passing to the limit consider small damping ($B_1 \ll 1$). The positive real part of the roots of the characteristic equation in the range $F_e < F < F_d$ for several small values of B_2 and, as an example, $B_1 = 0$ (i.e. $\beta = 0$) have been calculated and the results are displayed in Fig. 5, where F is plotted as a function of $\text{Re}\Omega$ for 9 values of B_2 . This Figure illustrates that for the larger values of B_2 , F_d represents the critical load because for $F > F_d$ some roots will have a non-vanishing positive real part. A small increase of the load above F_d will result in a large increase of this real part. For small values of B_2 , however, even though F_d is still strictly speaking the critical load, its significance is lessened, because a small increase of the load above F_d will not result any longer in a large increase of $\text{Re}\Omega$. Large increase of $\text{Re}\Omega$ will now be associated with small increase of a load which is slightly lower than F_e . For vanishing damping $\text{Re}\Omega = 0$ for any $F < F_e$. We thus conclude that during the limiting process the significance of F_d as a critical load is gradually transferred to F_e , and at the limit of vanishing damping ($B_1 \rightarrow 0$) F_e has to be considered as the critical load. It is apparent now that this conclusion could only be reached by considering the roots of the characteristic equation and not by merely applying the stability criteria of Routh-Hurwitz. Further, the reasons for the stability criteria yielding different critical loads for no damping and for vanishing damping can be better understood by having considered small damping.

Degree of Instability

It was established in the preceding section that for vanishing damping ($B_1 \rightarrow 0$) the four roots of the characteristic equation become identical to those of no damping ($B_1 = 0$) while the stability criteria alone would in general yield disparate critical loads in these two cases.

To establish a further connection between the mathematically derived critical loads for no damping ($B_1 = 0$) and vanishing damping ($B_1 \rightarrow 0$) it appears helpful to introduce into the discussion a concept which might be called "degree of instability" and which embodies a relaxation of the concept of instability as used when applying the kinetic stability criterion. According to this latter criterion a system is stable if a suitable disturbance results in a ^{bounded} motion in the vicinity of the equilibrium configuration, e.g., the system is unstable if a disturbance leads to oscillations with increasing amplitude (flutter instability). For this type of loss of stability one can state that from a practical point of view it will certainly matter how fast the amplitudes increase.

For example, should a suitable initial disturbance be merely doubled in a time interval which is large as compared to, say, some reference period, while the duration of the system being subjected to a nonconservative force is by comparison relatively short, the system may be considered practically stable, while, mathematically of course, one would have to conclude that it is unstable.

In order to weaken the kinetic stability criterion, one could prescribe arbitrarily the allowable increase of the disturbance and would then obtain for a given value of the load a critical time, not unlike

in the case of creep buckling. As an alternative, one could introduce another measure of the rate of amplitude increase. By analogy to decaying oscillations, where the logarithmic decrement serves the purpose of quantitatively assessing the rate of decay, we can use the same quantity also as a measure of the rate of amplitude increase. Thus

$$\delta = \log \frac{A_n}{A_{n+1}}$$

where A_n is the amplitude of the oscillation at a certain time t and A_{n+1} is the amplitude at $t + T$, where T is the period. In the present problem, neglecting the terms of decaying magnitude in the general solution of ϕ_1 , δ will generally be time-independent for flutter motions, except when the characteristic equation has equal pure imaginary roots.

The kinetic stability criterion requires $\delta \geq 0$ (i.e. $A_n \geq A_{n+1}$). A negative δ could properly be called the logarithmic increment and in a real system it is conceivable that δ may attain a certain value δ_c in a certain interval of time without the system losing its stability in any practical sense.

For $\beta = B_1/B_2 = 1$ the critical load F is displayed as a function of $B_1 = B_2 = B$ in Figs. 6 and 7. For however small but finite negative value of δ , the critical load for vanishing damping ($B \rightarrow 0$) will always be that for no damping ($B = 0$), namely F_e . However, the critical load for small damping ($B < 1$) may be smaller than F_e but for finite δ , however small, is always larger than F_d . For given δ the value of (small) damping B which is associated with the minimum value of the critical load can be determined.

For vanishing logarithmic increment ($\delta \rightarrow 0$) the function $\bar{r}(B)$ approaches a limiting curve which will contain the point F_d on the ordinate. For $\delta = 0$ the stability region is closed, i.e., points on the curve $\delta = 0$ in Fig. 7 are stable, including the point F_d on the ordinate. For $B = 0$ it is the point F_0 which separates stability from instability, but belongs itself to the instability region. This limiting process provides thus additional insight into the generation of the critical load F_d .

Continuous Cantilever

An attempt will be made now to interpret the results of the preceding sections, established with the aid of a simple two-degree-of-freedom model, as applied to a continuous cantilever beam, which represents possibly a more realistic system. This interpretation, however, is not without difficulty.

We shall assume that the internal damping of the continuous cantilever can be represented by Voigt elements, i.e., we use the Sezawa beam theory [8], and consider only the two lower modes of motion. The ordinary differential equation governing each mode X_i , of a cantilever with no force at the free end, is of the form

$$\ddot{X}_i + \frac{\gamma \omega_i^2}{E} \dot{X}_i + \omega_i^2 X_i = 0$$

where γ is the damping coefficient in the stress-strain relations and E is Young's modulus. The ratio of the damping coefficients of the first two modes is thus

$$\bar{\beta} = \frac{\omega_1^2}{\omega_2^2} = \left(\frac{1.875}{4.694} \right)^4 = 0.0256$$

This ratio for the continuous cantilever should now be compared with that of the cantilever model and for this purpose one should uncouple the two equations governing the model. It is known [9], however, that whenever a dissipation energy is accounted for, in addition to kinetic and potential energies, such uncoupling can, in general, not be effected and this is the difficulty alluded to above. In the system under consideration uncoupling becomes possible in the special case given by $\beta = 1$ (i.e., $b_1 = b_2 = b$) because in this case the dissipation function becomes proportional to the potential energy.

The transformation

$$\varphi_1 = \frac{1}{\sqrt{2}} y_1$$

$$\varphi_2 = -\frac{1}{\sqrt{2}} y_1 + y_2$$

with

$$y_1 = \xi_1 \cos \theta - \xi_2 \sin \theta$$

$$y_2 = \xi_1 \sin \theta + \xi_2 \cos \theta$$

leads to the uncoupled equations

$$\ddot{\xi}_1 + \frac{b}{4ml^2} (7 - \sqrt{41}) \dot{\xi}_1 + \frac{c}{4ml^2} (7 - \sqrt{41}) \xi_1 = 0$$

$$\ddot{\xi}_2 + \frac{b}{4ml^2} (7 + \sqrt{41}) \dot{\xi}_2 + \frac{c}{4ml^2} (7 + \sqrt{41}) \xi_2 = 0$$

In this representation the ratio of the damping coefficients is given by

$$\bar{\beta} = \frac{7 - \sqrt{41}}{7 + \sqrt{41}} = 0.0446$$

The $\bar{\beta}$ and β are relatively close and one can conclude that in the original coordinates φ_i the ratio of the damping coefficients β has to be taken in the vicinity of unity to correspond to the continuous cantilever.

Further, for many structural materials the fraction of critical damping $\epsilon_1 = \frac{\gamma \omega_1}{2E}$ is known to be of the order of 10^{-3} .

Since (with $\beta = 1$)

$$2\epsilon_1 \omega_1 = \frac{b}{4m\ell^2} (7 - \sqrt{41})$$

and

$$\omega_1 = \frac{\sqrt{c}}{2\ell \sqrt{m}} \sqrt{(7 - \sqrt{41})}$$

the fraction of critical damping in the first mode will be

$$\epsilon_1 = B \sqrt{7 - \sqrt{41}} = 0.775 B$$

Similarly, for the second mode it will be

$$\epsilon_2 = 3.661 B$$

Thus B is of the same order of magnitude as ϵ_1 , i.e. 10^{-3} , and damping will have indeed a destabilizing effect, as seen from Fig. 6.

Concluding Remarks

An examination of the roots of the characteristic equation and the introduction of the concept of degree of instability make it possible to establish a generic relationship between the critical loads for no damping and for small and vanishing damping. Routh-Hurwitz criteria alone proved to be insufficient to determine the critical load for

vanishing damping, which is the same as for no damping. It is small damping, rather than vanishing damping, which is responsible for the destabilizing effect. The strong dependence of the critical load on the ratio of the damping coefficients, however, leaves a requirement for further investigation, which should include other damping mechanism, effects of nonlinearity and different types of nonconservative forces.

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CAPTIONS OF FIGURES

1. Two-degree-of-freedom model
2. Orthographic projections and the perspective of the root curves of the characteristic equation with no damping
3. Orthographic projections and the perspective of the root curves of the characteristic equation with damping
4. Critical load versus ratio of damping coefficients for $B_1 \ll 1$
5. Significance of the critical load F_d as B_2 increases
6. Critical load for various degrees of instability versus small damping coefficients
7. Critical load for various degrees of instability versus large damping coefficients.

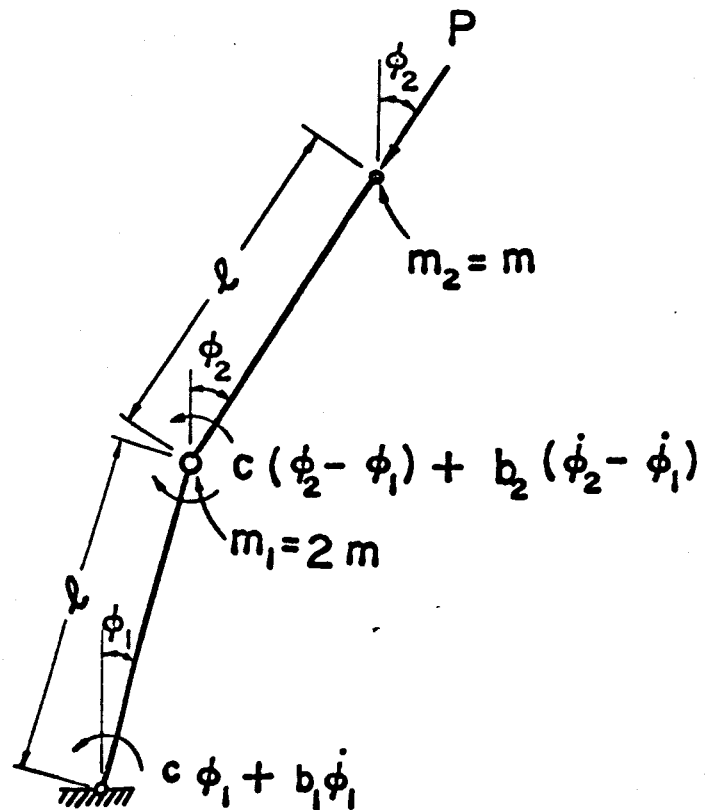


Fig. 1

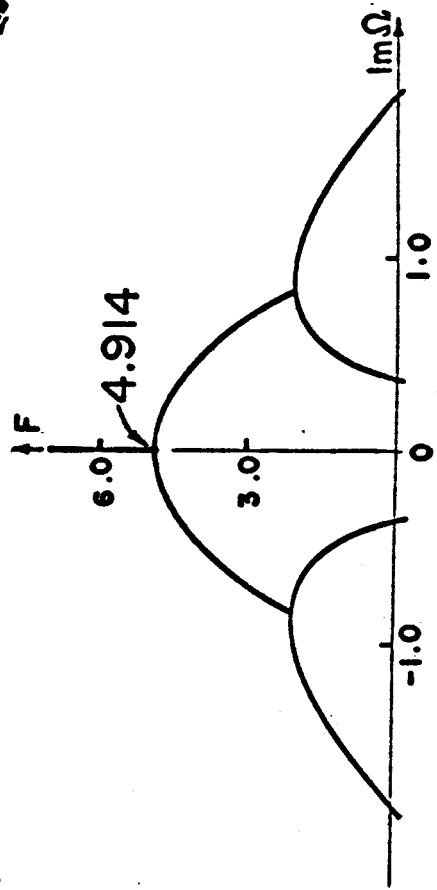
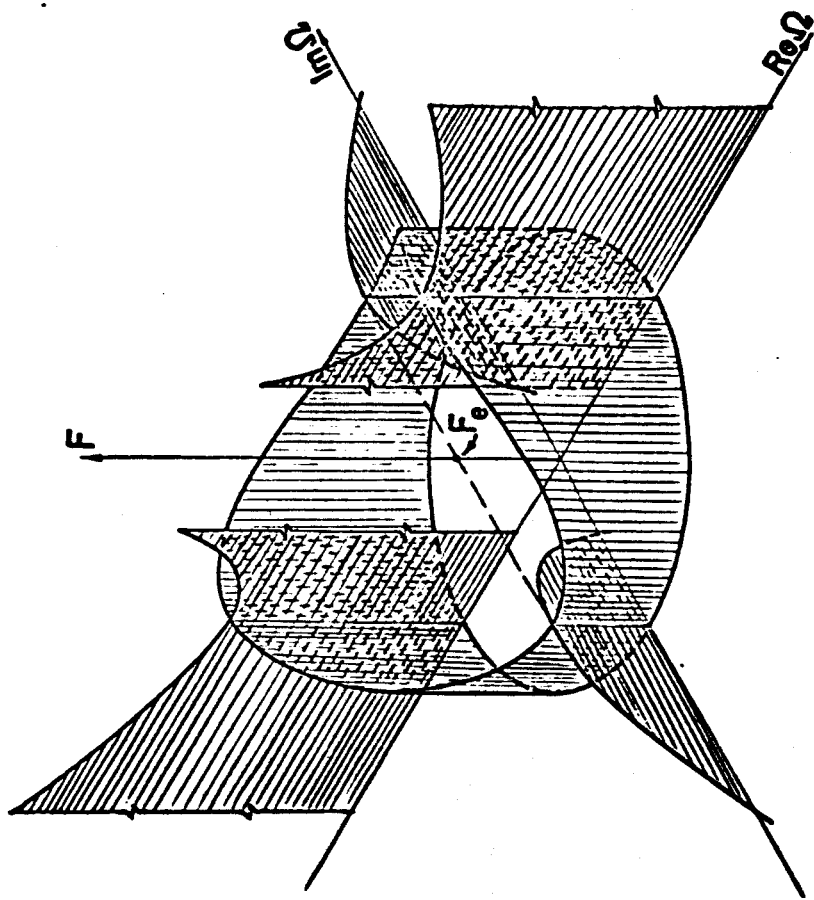
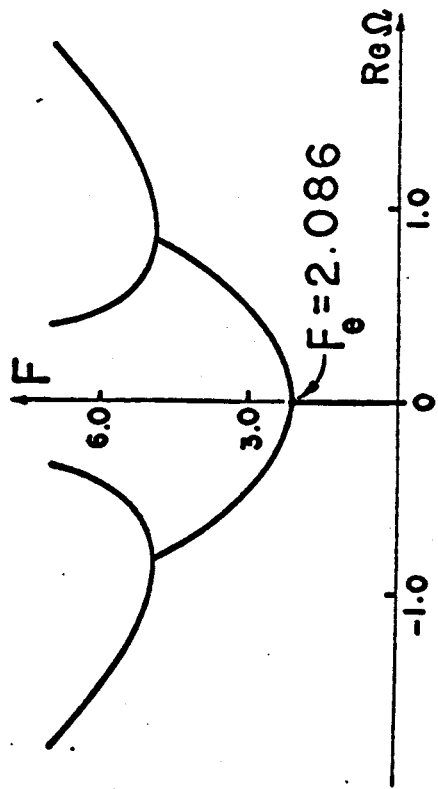
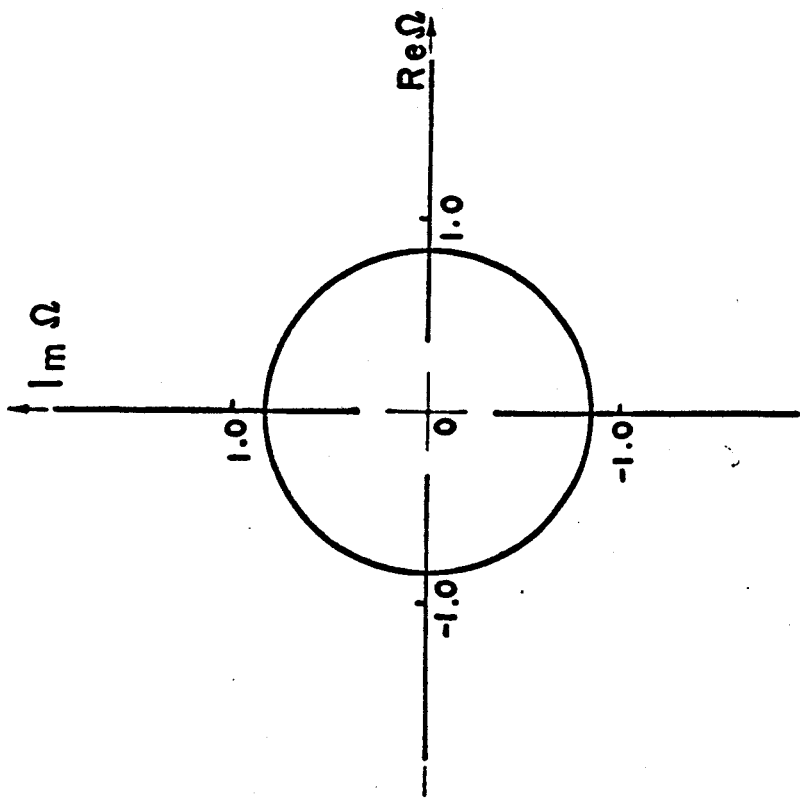


Fig. 2

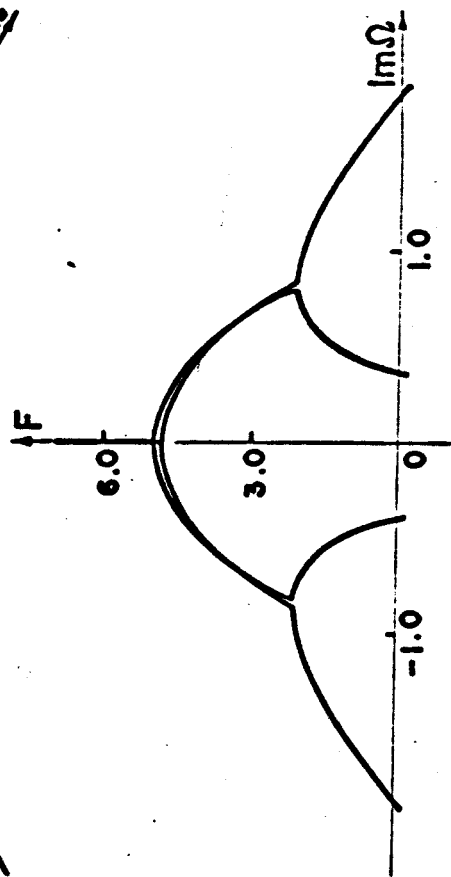
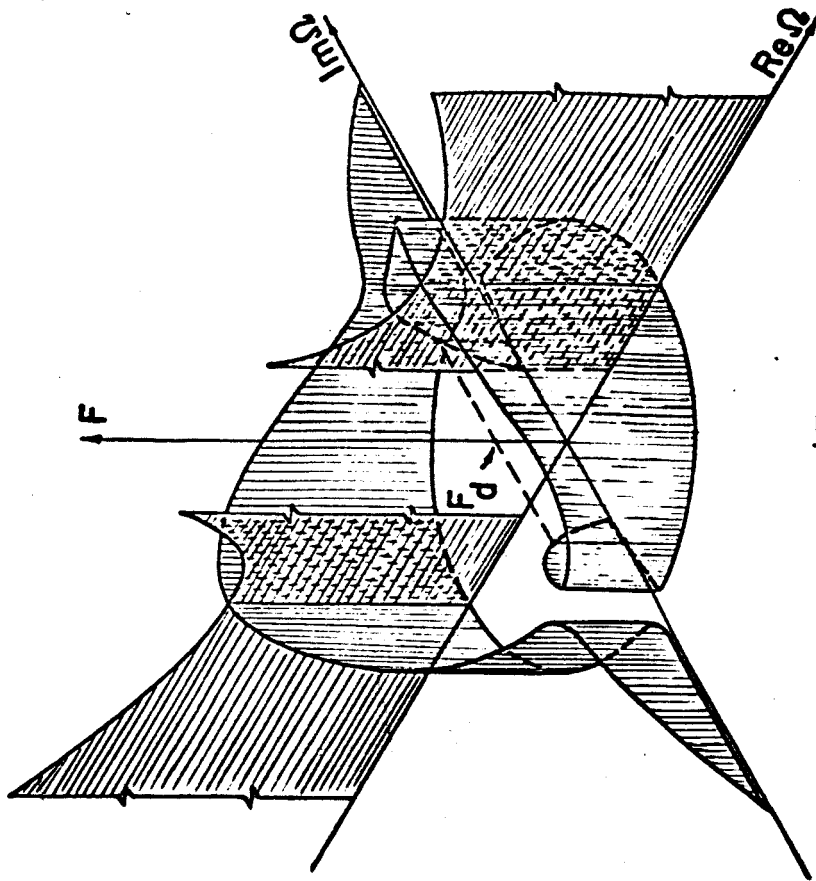
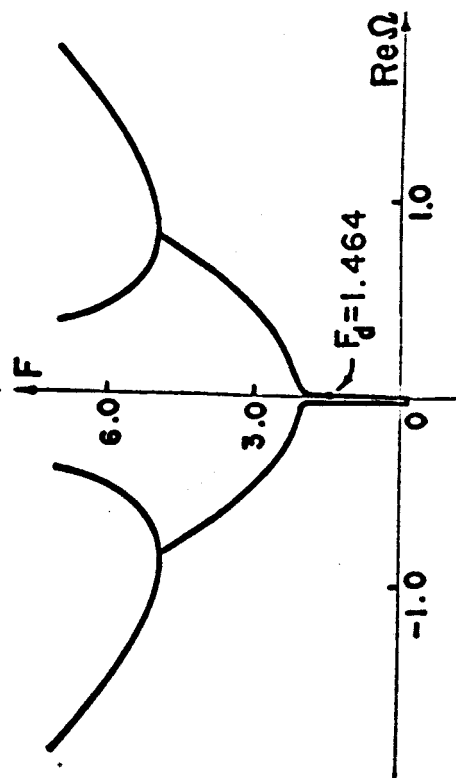
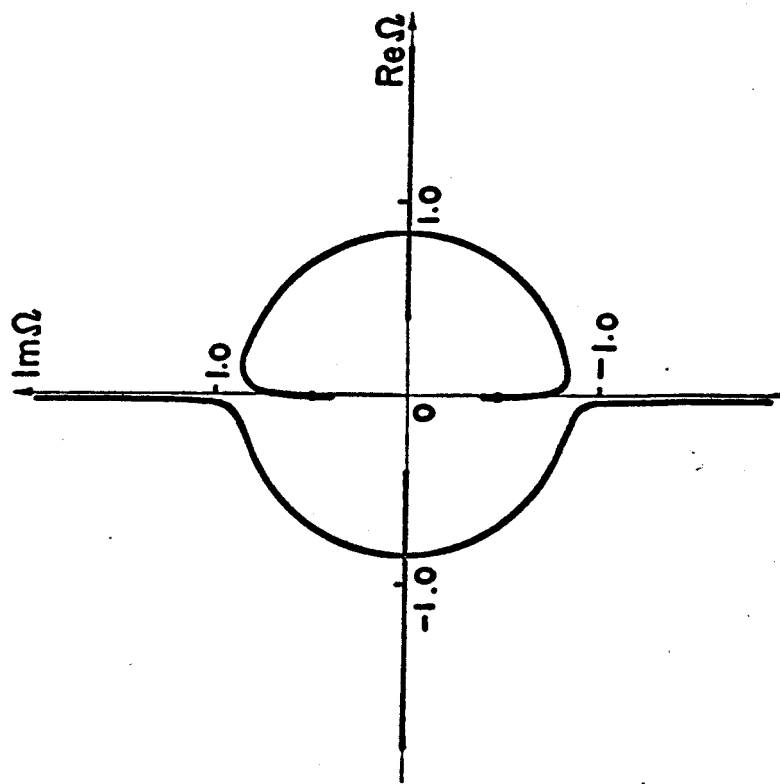


Fig. 3

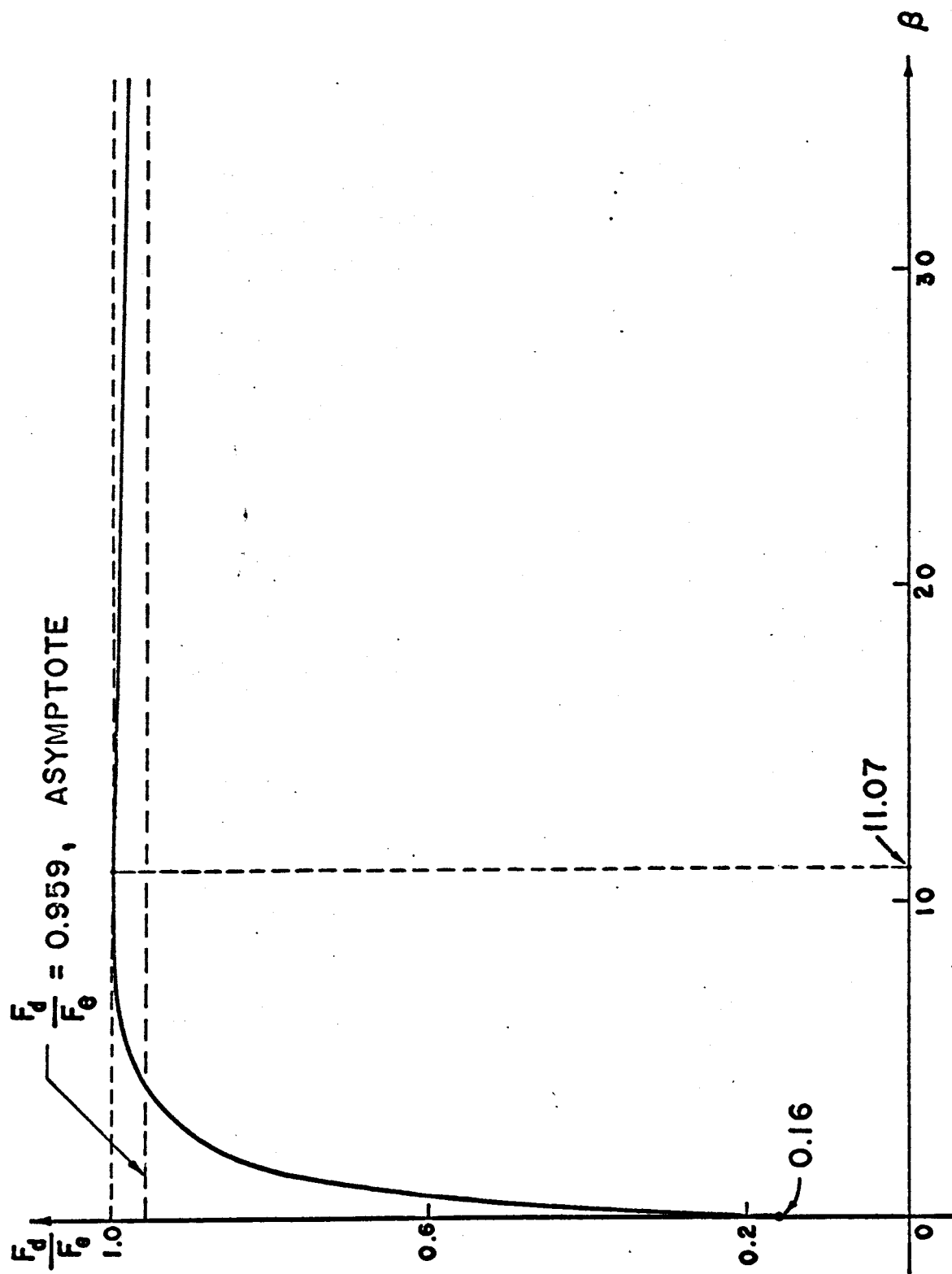


Fig. 4

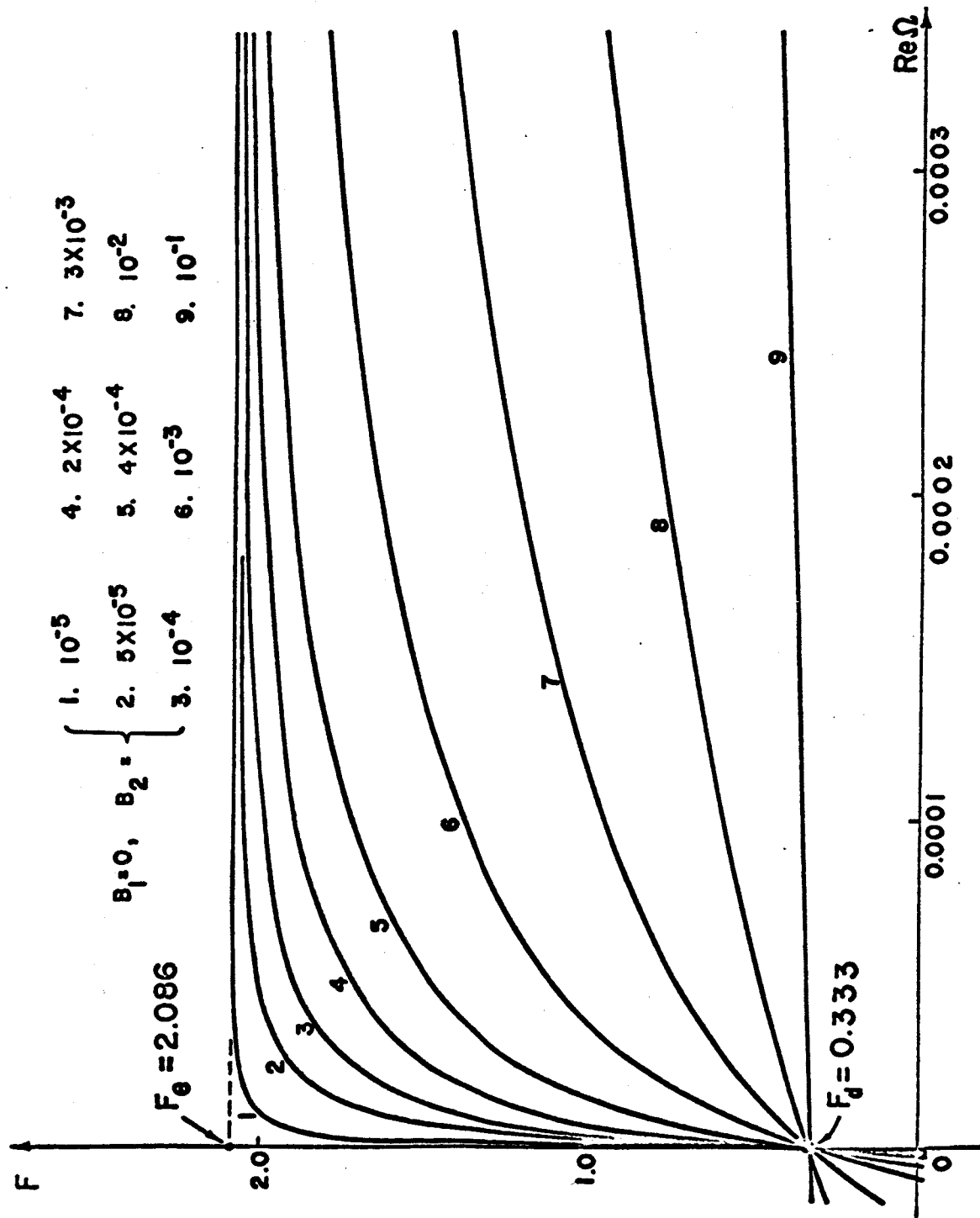


Fig. 5

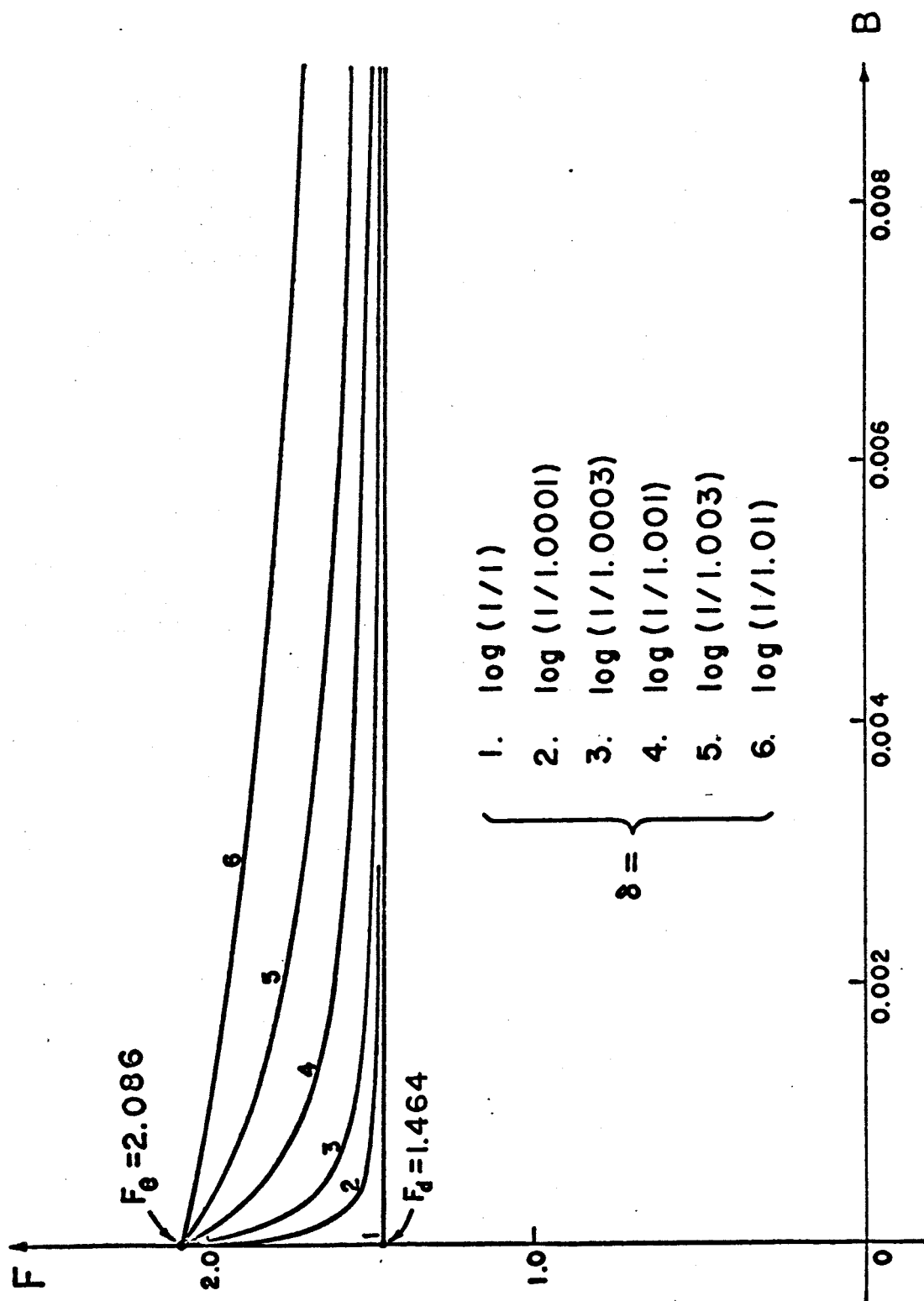


Fig. 6

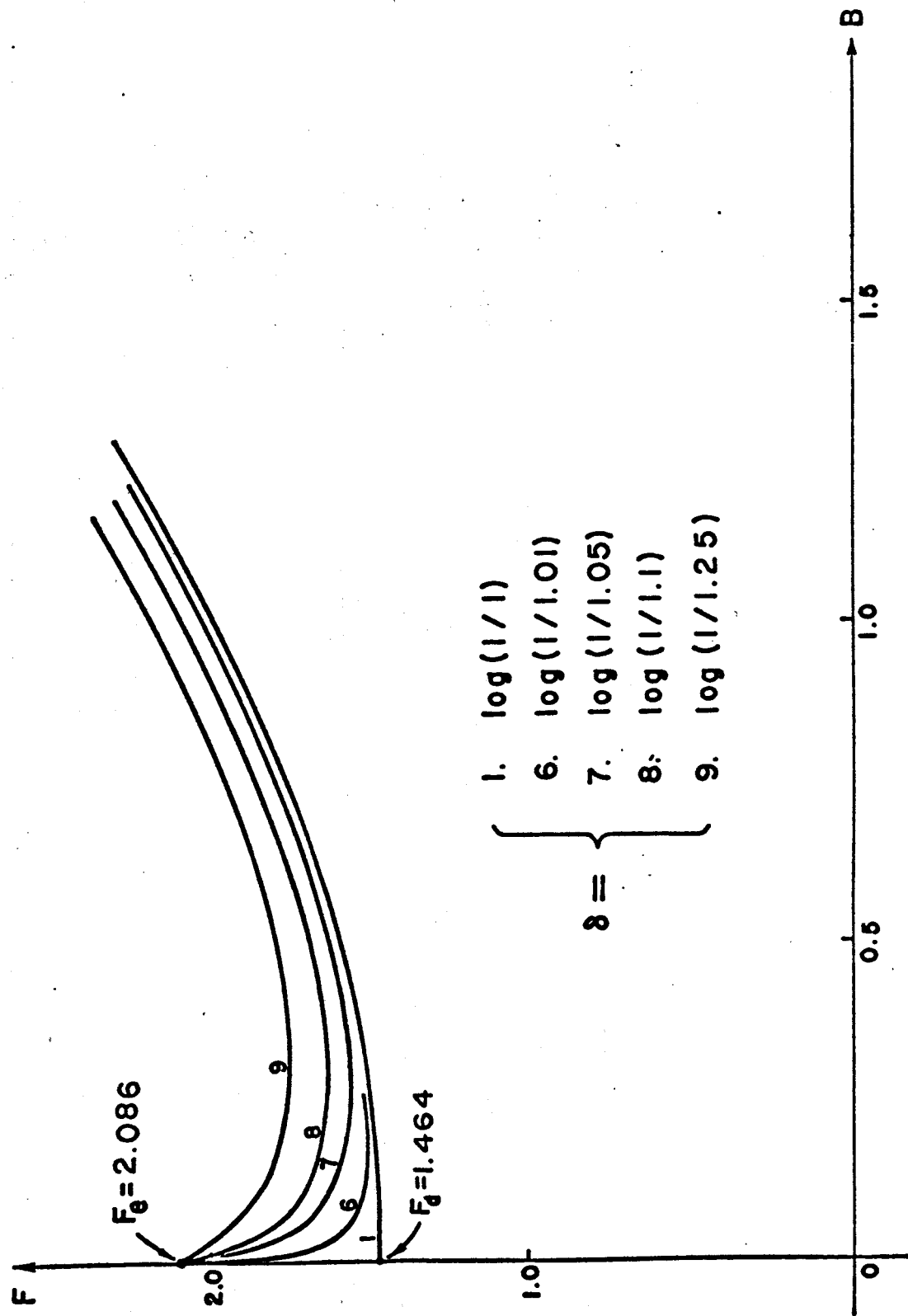


Fig. 7

SUMMARY

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It has been discovered by Ziegler [1] a little more than a decade ago, that internal damping may have a destabilizing effect in nonconservative elastic system. He considered a double pendulum with viscoelastic hinges as a model of an elastic bar with internal damping and let a tangential force act at the free end. The critical loading obtained in complete absence of damping was found to be considerably higher than by including damping at the outset of the analysis and then letting the damping coefficients approach zero (vanishing damping) in the expression for the critical force.

This rather surprising and paradoxical finding was ascribed in later studies by Ziegler [2], [3] to the possibility that internal damping is inadequately represented by linear damping forces which are linear combinations of the generalized velocities and that the hysteresis effect should be taken into account.

The destabilizing effect of damping was further elaborated upon by Bolotin [4] who considered a general two-degree-of-freedom system not related to any particular mechanical model and who found additionally that the destabilizing effect in the presence of slight and vanishing damping is highly dependent on the relative magnitude of damping coefficients in the two degrees freedom.

It is the aim of the present investigation to make an attempt at supplying some additional insight into the destabilizing effects of linear velocity-dependent damping in nonconservative systems, without raising the question here as to the suitability of this damping mechanism

for a realistic system. For this purpose the system discussed by Ziegler is reconsidered, and not only the stability conditions are investigated but also the roots of the characteristic equations themselves. Plots of these roots for various ranges of loading illustrate graphically how the paradoxical effects of vanishing damping are generated. Further, the results of the mathematical stability investigations are interpreted in physical terms by introducing the concept of degree of instability.

These concepts permit to carry out a gradual transition from the case of small damping to the case of vanishing damping and relate them to the case of no damping. Finally, some remarks are made with regard to possible behavior of an elastic bar with distributed parameters.

The Model

We consider a double pendulum, Fig. 1, composed of two rigid weightless bars of equal length l , which carry concentrated masses $m_1 = 2m$, $m_2 = m$. The generalized coordinates φ_1 , φ_2 are taken to be small. A load P applied at the free end is assumed to be acting at an angle φ_2 (follower force). At the hinges the restoring moments $c\varphi_1 + b_1\dot{\varphi}_1$ and $c(\varphi_2 - \varphi_1) + b_2(\dot{\varphi}_2 - \dot{\varphi}_1)$ are induced.

The kinetic energy T , the dissipation function D , the potential energy V , and the generalized forces Q_1 and Q_2 are:

$$T = \frac{1}{2} m l^2 (3\dot{\varphi}_1^2 + 2\dot{\varphi}_1\dot{\varphi}_2 + \dot{\varphi}_2^2)$$

$$D = \frac{1}{2} b_1 \dot{\varphi}_1^2 + \frac{1}{2} b_2 (\dot{\varphi}_1^2 - 2\dot{\varphi}_1\dot{\varphi}_2 + \dot{\varphi}_2^2)$$

$$V = \frac{1}{2} c (2\varphi_1^2 - 2\varphi_1\varphi_2 + \varphi_2^2)$$

$$Q_1 = Pl (\varphi_1 - \varphi_2)$$

$$Q_2 = 0$$

Lagrange's equations in the form

$$\frac{d}{dt} \left(\frac{\partial T}{\partial \dot{\varphi}_i} \right) + \frac{\partial D}{\partial \dot{\varphi}_i} - \frac{\partial T}{\partial \varphi_i} + \frac{\partial V}{\partial \varphi_i} = Q_i \quad (i = 1, 2)$$

are employed to establish the linear equations of motion

$$3ml^2\ddot{\varphi}_1 + (b_1 + b_2)\dot{\varphi}_1 - (Pl - 2c)\varphi_1 + ml^2\ddot{\varphi}_2 - b_2\dot{\varphi}_2 + (Pl - c)\varphi_2 = 0$$

$$ml^2\ddot{\varphi}_1 - b_2\dot{\varphi}_1 - c\varphi_1 + ml^2\ddot{\varphi}_2 + b_2\dot{\varphi}_2 + c\varphi_2 = 0$$

which, upon stipulating solutions of the form

$$\varphi_i = A_i e^{\omega t} \quad (i = 1, 2)$$

yield the characteristic equation

$$p_0 \omega^4 + p_1 \omega^3 + p_2 \omega^2 + p_3 \omega + p_4 = 0$$

with the coefficients

$$p_0 = 2$$

$$p_1 = B_1 + 6B_2$$

$$p_2 = 7 - 2F + B_1 B_2$$

$$p_3 = B_1 + B_2$$

$$p_4 = 1$$

and the dimensionless quantities

$$\Omega = l(\frac{m}{c})^{\frac{1}{2}} \omega$$

$$B_i = \frac{b_i}{l(\text{cm})^{\frac{1}{2}}} \quad (i = 1, 2)$$

$$F = \frac{Pl}{c}$$

In the absence of damping ($B_1 = B_2 = 0$), the characteristic equation is a biquadratic

$$2\Omega^4 + (7-2F)\Omega^2 + 1 = 0$$

Critical Loads

From the assumed form of the time-dependence for the coordinates φ_i and on the basis of the kinetic stability criterion, it is evident that if all four roots of the characteristic equation are distinct, the necessary and sufficient conditions for stability are that the real roots and the real parts of the complex roots should be all negative or zero. In case of equal roots the general solution of φ_i will have terms which contain powers of t as a factor. If the real parts of equal roots are negative, the system will be stable (vibration with decreasing amplitude), but if these real parts are zero or positive, stability will not exist (vibration with increasing amplitude).

Turning our attention first to the case of an initially undamped system, the four roots of the biquadratic equation as a function of F are

$$\Omega_{1,2,3,4} = \frac{1}{2} \left[\pm \sqrt{F - (7/2 - \sqrt{2})} \pm \sqrt{F - (7/2 + \sqrt{2})} \right]$$

and are plotted in Fig. 2. The projections of the root curves on the

real plane ($\text{Im}\Omega = 0$), the imaginary plane ($\text{Re}\Omega = 0$) and the complex plane ($F = 0$) are also shown in Fig. 2.

It is found that there will always be two roots with positive real part if $F > \frac{7}{2} - \sqrt{2} = 2.086 = F_e$. For $F = F_e$ there exist two pairs of equal roots whose real parts are all zero. Thus the system is unstable for $F \geq F_e$. For $F < F_e$ all roots are distinct and pure imaginary and thus the system is stable. A further discussion of an initially undamped system is presented in [5].

We consider next a slightly damped system, assuming $B_1 = B_2 = 0.01$. No simple expressions for the four roots of the quartic equation exist; the numerical results obtained are illustrated in Fig. 3 where a perspective view is supplemented by three projections on the same three planes as in Fig. 2. Two roots will have a positive real part for $F > 1.464 = F_d$.

Stability can be investigated directly without determining the roots of the characteristic equation by applying the Routh-Hurwitz criteria [6], which require that all coefficients p_j ($j = 0, \dots, 4$) of the characteristic equation and the quantities

$$S = p_1 p_2 - p_0 p_3$$

$$X = p_1 p_2 p_3 - p_0 p_3^2 - p_1^2 p_4$$

be positive. For positive damping these stability conditions are satisfied, provided

$$p_2 = 2[-F + \frac{1}{2}(7 + B_1 B_2)] > 0$$

$$S = 2(B_1 + 6B_2) \left\{ -F + \left[\frac{5(B_1 + 8B_2)}{2(B_1 + 6B_2)} + \frac{1}{2} B_1 B_2 \right] \right\} > 0$$

$$X = 2(B_1^2 + 7B_1 B_2 + 6B_2^2) \left\{ -F + \left[\frac{4B_1^2 + 33B_1 B_2 + 4B_2^2}{2(B_1^2 + 7B_1 B_2 + 6B_2^2)} + \frac{1}{2} B_1 B_2 \right] \right\} > 0$$

For the system to be stable F must satisfy the following three inequalities, where $\beta = B_1/B_2$, $0 \leq \beta \leq \infty$

$$F < \frac{7}{2} + \frac{1}{2} B_1 B_2$$

$$F < \frac{5(\beta+8)}{2(\beta+6)} + \frac{1}{2} B_1 B_2$$

$$F < \frac{4\beta^2 + 33\beta + 4}{2(\beta^2 + 7\beta + 6)} + \frac{1}{2} B_1 B_2$$

Since

$$\frac{5(\beta+8)}{2(\beta+6)} = \frac{5}{2} + \frac{5}{\beta+6} \leq \frac{10}{3} < \frac{7}{2}$$

$$\frac{4\beta^2 + 33\beta + 4}{2(\beta^2 + 7\beta + 6)} = \frac{5(\beta+8)}{2(\beta+6)} - \frac{\beta+3}{2(\beta+1)} < \frac{5(\beta+8)}{2(\beta+6)} < \frac{7}{2}$$

for whatever β in its range, it is evident that the critical load will be governed by the third inequality, i.e.,

$$\bar{F}_d = \frac{4\beta^2 + 33\beta + 4}{2(\beta^2 + 7\beta + 6)} + \frac{1}{2} B_1 B_2$$

which depends on the ratio as well as the magnitudes of the damping coefficients.

For $B_1 \ll 1$, as well as in the limit of vanishing damping, \bar{F}_d becomes

$$F_d = \frac{4\beta^2 + 33\beta + 4}{2(\beta^2 + 7\beta + 6)}$$

which is highly dependent on β and is in general smaller but never larger than F_e . The ratio of F_d to F_e versus β is plotted in Fig. 4. It is noted that when $\beta = 4 + 5\sqrt{2} = 11.07$, F_d/F_e reaches its maximum value 1. The destabilizing effect is thus eliminated in this particular case, similar to that found by Bolotin [4]. For $\beta = 0$, F_d/F_e reaches its minimum value 0.16; i.e., the maximum destabilizing effect is about 84% in the present two-degree-of-freedom system.

The Case of Vanishing Damping

The two disparate values of the critical load for no damping ($B_1 = 0$) and vanishing damping ($B_1 \rightarrow 0$) justify a more detailed investigation of the limiting process as the damping coefficients approach zero.

Let us examine first the limiting process for the roots of the characteristic equation. It can be shown with the aid of the theory of equations [7] that if $B_1 \ll 1$ and $F < 4.914$ this equation will have four complex roots. Let these roots be

$$\Omega = \begin{cases} \gamma_1 \pm i\gamma_2 \\ \lambda_1 \pm i\lambda_2 \end{cases}$$

Then one can write [6], [7]

$$2(\gamma_1 + \lambda_1) = -\frac{p_1}{p_0}$$

$$4\gamma_1\lambda_1 [(\gamma_1+\lambda_1)^2 + (\gamma_2+\lambda_2)^2][(\gamma_1+\lambda_1)^2 + (\gamma_2-\lambda_2)^2] = \frac{\chi}{p_0^3}$$

where p_0 , p_1 and χ are as defined earlier. For vanishing damping

$$\begin{cases} \gamma_1 + \lambda_1 = 0 \\ \gamma_1\lambda_1 [(\gamma_1+\lambda_1)^2 + (\gamma_2+\lambda_2)^2][(\gamma_1+\lambda_1)^2 + (\gamma_2-\lambda_2)^2] = 0 \end{cases}$$

Hence $\gamma_1 = -\lambda_1, \gamma_2 = \lambda_2$

or $\gamma_1 = \lambda_1 = 0$

Thus

$$\Omega = \begin{cases} \pm i\gamma_2 \\ \pm i\lambda_2 \end{cases} \quad \text{or} \quad \Omega = \begin{cases} \gamma_1 \pm i\gamma_2 \\ -\gamma_1 \pm i\gamma_2 \end{cases}$$

and a substitution of these four roots into the characteristic equation will show that they are the same as in the case of no damping.

In the case of $F \geq 4.914$, the four roots will be all real for small B_1 . Let

$$\Omega = \begin{cases} u_1 \pm u_2 \\ v_1 \pm v_2 \end{cases}$$

In the limit of vanishing damping one can show similarly that either $u_1 = v_1 = 0$ or $u_1 = -v_1$ and $u_2 = v_2$. For either alternative, substitution into the characteristic equation reveals that the roots are the same as in the case of no damping.

Thus the conclusion is reached that whatever F the roots of the characteristic equation for no initial damping ($B_1 = 0$) are identical to those of vanishing damping ($B_1 \rightarrow 0$). This implies that the motions

of the system, for some given initial conditions, and whatever F , will be identical in the case of no damping ($B_1 = 0$) and vanishing damping ($B_1 \rightarrow 0$).

We focus attention next on the loading F in the two cases and before passing to the limit consider small damping ($B_1 \ll 1$). The positive real part of the roots of the characteristic equation in the range $F_e < F < F_d$ for several small values of B_2 and, as an example, $B_1 = 0$ (i.e. $\beta = 0$) have been calculated and the results are displayed in Fig. 5, where F is plotted as a function of $\text{Re}\Omega$ for 9 values of B_2 . This Figure illustrates that for the larger values of B_2 , F_d represents the critical load because for $F > F_d$ some roots will have a non-vanishing positive real part. A small increase of the load above F_d will result in a large increase of this real part. For small values of B_2 , however, even though F_d is still strictly speaking the critical load, its significance is lessened, because a small increase of the load above F_d will not result any longer in a large increase of $\text{Re}\Omega$. Large increase of $\text{Re}\Omega$ will now be associated with small increase of a load which is slightly lower than F_e . For vanishing damping $\text{Re}\Omega = 0$ for any $F < F_e$. We thus conclude that during the limiting process the significance of F_d as a critical load is gradually transferred to F_e , and at the limit of vanishing damping ($B_1 \rightarrow 0$) F_e has to be considered as the critical load. It is apparent now that this conclusion could only be reached by considering the roots of the characteristic equation and not by merely applying the stability criteria of Routh-Hurwitz. Further, the reasons for the stability criteria yielding different critical loads for no damping and for vanishing damping can be better understood by having considered small damping.

Degree of Instability

It was established in the preceding section that for vanishing damping ($B_1 \rightarrow 0$) the four roots of the characteristic equation become identical to those of no damping ($B_1 = 0$) while the stability criteria alone would in general yield disparate critical loads in these two cases.

To establish a further connection between the mathematically derived critical loads for no damping ($B_1 = 0$) and vanishing damping ($B_1 \rightarrow 0$) it appears helpful to introduce into the discussion a concept which might be called "degree of instability" and which embodies a relaxation of the concept of instability as used when applying the kinetic stability criterion. According to this latter criterion a system is stable if a suitable disturbance results in a ^{bounded} motion in the vicinity of the equilibrium configuration, e.g., the system is unstable if a disturbance leads to oscillations with increasing amplitude (flutter instability). For this type of loss of stability one can state that from a practical point of view it will certainly matter how fast the amplitudes increase.

For example, should a suitable initial disturbance be merely doubled in a time interval which is large as compared to, say, some reference period, while the duration of the system being subjected to a nonconservative force is by comparison relatively short, the system may be considered practically stable, while, mathematically of course, one would have to conclude that it is unstable.

In order to weaken the kinetic stability criterion, one could prescribe arbitrarily the allowable increase of the disturbance and would then obtain for a given value of the load a critical time, not unlike

in the case of creep buckling. As an alternative, one could introduce another measure of the rate of amplitude increase. By analogy to decaying oscillations, where the logarithmic decrement serves the purpose of quantitatively assessing the rate of decay, we can use the same quantity also as a measure of the rate of amplitude increase. Thus

$$\delta = \log \frac{A_n}{A_{n+1}}$$

where A_n is the amplitude of the oscillation at a certain time t and A_{n+1} is the amplitude at $t + T$, where T is the period. In the present problem, neglecting the terms of decaying magnitude in the general solution of φ_i , δ will generally be time-independent for flutter motions, except when the characteristic equation has equal pure imaginary roots.

The kinetic stability criterion requires $\delta \geq 0$ (i.e. $A_n \geq A_{n+1}$). A negative δ could properly be called the logarithmic increment and in a real system it is conceivable that δ may attain a certain value δ_c in a certain interval of time without the system losing its stability in any practical sense.

For $\beta = B_1/B_2 = 1$ the critical load F is displayed as a function of $B_1 = B_2 = B$ in Figs. 6 and 7. For however small but finite negative value of δ , the critical load for vanishing damping ($B \rightarrow 0$) will always be that for no damping ($B = 0$), namely F_0 . However, the critical load for small damping ($B < 1$) may be smaller than F_0 but for finite δ , however small, is always larger than F_0 . For given δ the value of (small) damping B which is associated with the minimum value of the critical load can be determined.

For vanishing logarithmic increment ($\delta \rightarrow 0$) the function $\bar{F}(B)$ approaches a limiting curve which will contain the point F_d on the ordinate. For $\delta = 0$ the stability region is closed, i.e., points on the curve $\delta = 0$ in Fig. 7 are stable, including the point F_d on the ordinate. For $B = 0$ it is the point F_0 which separates stability from instability, but belongs itself to the instability region. This limiting process provides thus additional insight into the generation of the critical load F_d .

Continuous Cantilever

An attempt will be made now to interpret the results of the preceding sections, established with the aid of a simple two-degree-of-freedom model, as applied to a continuous cantilever beam, which represents possibly a more realistic system. This interpretation, however, is not without difficulty.

We shall assume that the internal damping of the continuous cantilever can be represented by Voigt elements, i.e., we use the Sezawa beam theory [8], and consider only the two lower modes of motion. The ordinary differential equation governing each mode X_i , of a cantilever with no force at the free end, is of the form

$$\ddot{X}_i + \frac{\gamma \omega_i^2}{E} \dot{X}_i + \omega_i^2 X_i = 0$$

where γ is the damping coefficient in the stress-strain relations and E is Young's modulus. The ratio of the damping coefficients of the first two modes is thus

$$\bar{\beta} = \frac{\omega_1^2}{\omega_2^2} = \left(\frac{1.875}{4.694} \right)^4 = 0.0256$$

This ratio for the continuous cantilever should now be compared with that of the cantilever model and for this purpose one should uncouple the two equations governing the model. It is known [9], however, that whenever a dissipation energy is accounted for, in addition to kinetic and potential energies, such uncoupling can, in general, not be effected and this is the difficulty alluded to above. In the system under consideration uncoupling becomes possible in the special case given by $\beta = 1$ (i.e., $b_1 = b_2 = b$) because in this case the dissipation function becomes proportional to the potential energy.

The transformation

$$\varphi_1 = \frac{1}{\sqrt{2}} y_1$$

$$\varphi_2 = -\frac{1}{\sqrt{2}} y_1 + y_2$$

with

$$y_1 = \xi_1 \cos \theta - \xi_2 \sin \theta$$

$$y_2 = \xi_1 \sin \theta + \xi_2 \cos \theta$$

leads to the uncoupled equations

$$\ddot{\xi}_1 + \frac{b}{4ml^2} (7 - \sqrt{41}) \dot{\xi}_1 + \frac{c}{4ml^2} (7 - \sqrt{41}) \xi_1 = 0$$

$$\ddot{\xi}_2 + \frac{b}{4ml^2} (7 + \sqrt{41}) \dot{\xi}_2 + \frac{c}{4ml^2} (7 + \sqrt{41}) \xi_2 = 0$$

In this representation the ratio of the damping coefficients is given by

$$\bar{\beta} = \frac{7 - \sqrt{41}}{7 + \sqrt{41}} = 0.0446$$

The $\bar{\beta}$ and $\bar{\beta}^*$ are relatively close and one can conclude that in the original coordinates φ_i the ratio of the damping coefficients β has to be taken in the vicinity of unity to correspond to the continuous cantilever.

Further, for many structural materials the fraction of critical damping $\epsilon_i = \frac{\gamma \omega_i}{2E}$ is known to be of the order of 10^{-3} .

Since (with $\beta = 1$)

$$2\epsilon_1 \omega_1 = \frac{b}{4ml^2} (7 - \sqrt{41})$$

and

$$\omega_1 = \frac{\sqrt{c}}{2l\sqrt{m}} \sqrt{(7 - \sqrt{41})}$$

the fraction of critical damping in the first mode will be

$$\epsilon_1 = B \sqrt{7 - \sqrt{41}} = 0.775 B$$

Similarly, for the second mode it will be

$$\epsilon_2 = 3.661 B$$

Thus B is of the same order of magnitude as ϵ_1 , i.e. 10^{-3} , and damping will have indeed a destabilizing effect, as seen from Fig. 6.

Concluding Remarks

An examination of the roots of the characteristic equation and the introduction of the concept of degree of instability make it possible to establish a generic relationship between the critical loads for no damping and for small and vanishing damping. Routh-Hurwitz criteria alone proved to be insufficient to determine the critical load for

vanishing damping, which is the same as for no damping. It is small damping, rather than vanishing damping, which is responsible for the destabilizing effect. The strong dependence of the critical load on the ratio of the damping coefficients, however, leaves a requirement for further investigation, which should include other damping mechanism, effects of nonlinearity and different types of nonconservative forces.

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CAPTIONS OF FIGURES

1. Two-degree-of-freedom model
2. Orthographic projections and the perspective of the root curves of the characteristic equation with no damping
3. Orthographic projections and the perspective of the root curves of the characteristic equation with damping
4. Critical load versus ratio of damping coefficients for $B_1 \ll 1$
5. Significance of the critical load F_d as B_2 increases
6. Critical load for various degrees of instability versus small damping coefficients
7. Critical load for various degrees of instability versus large damping coefficients.

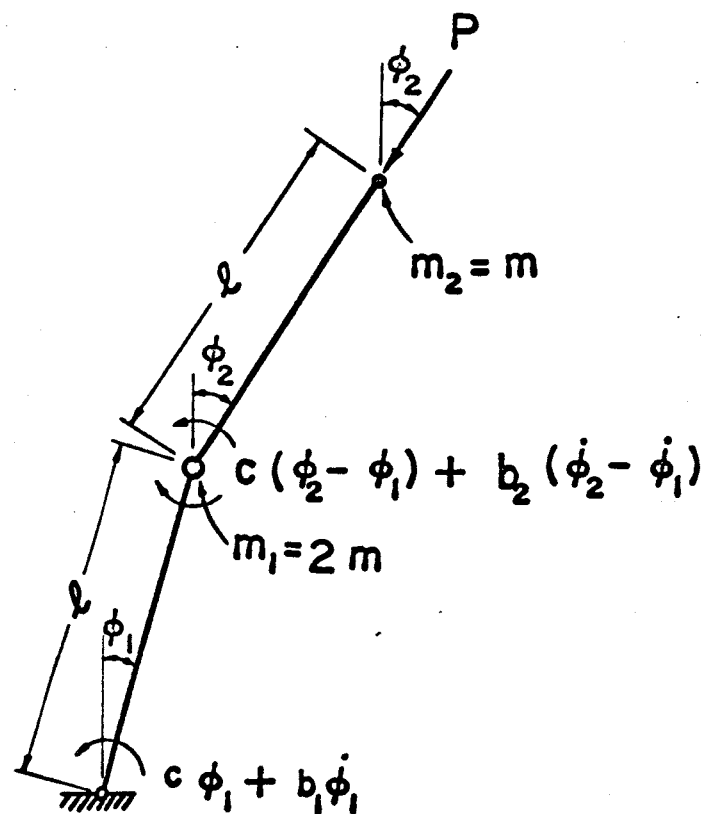


Fig. 1

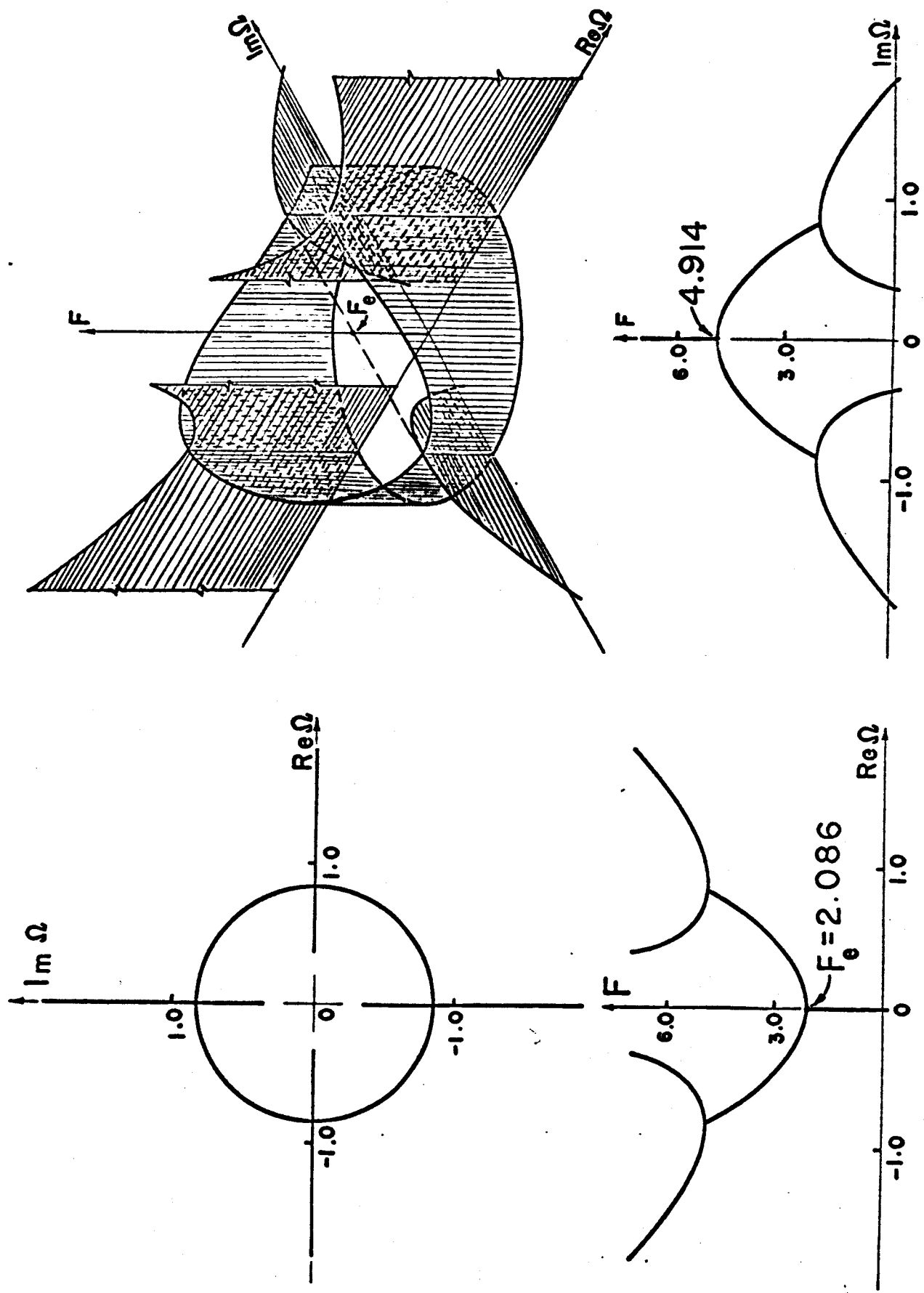


Fig. 2

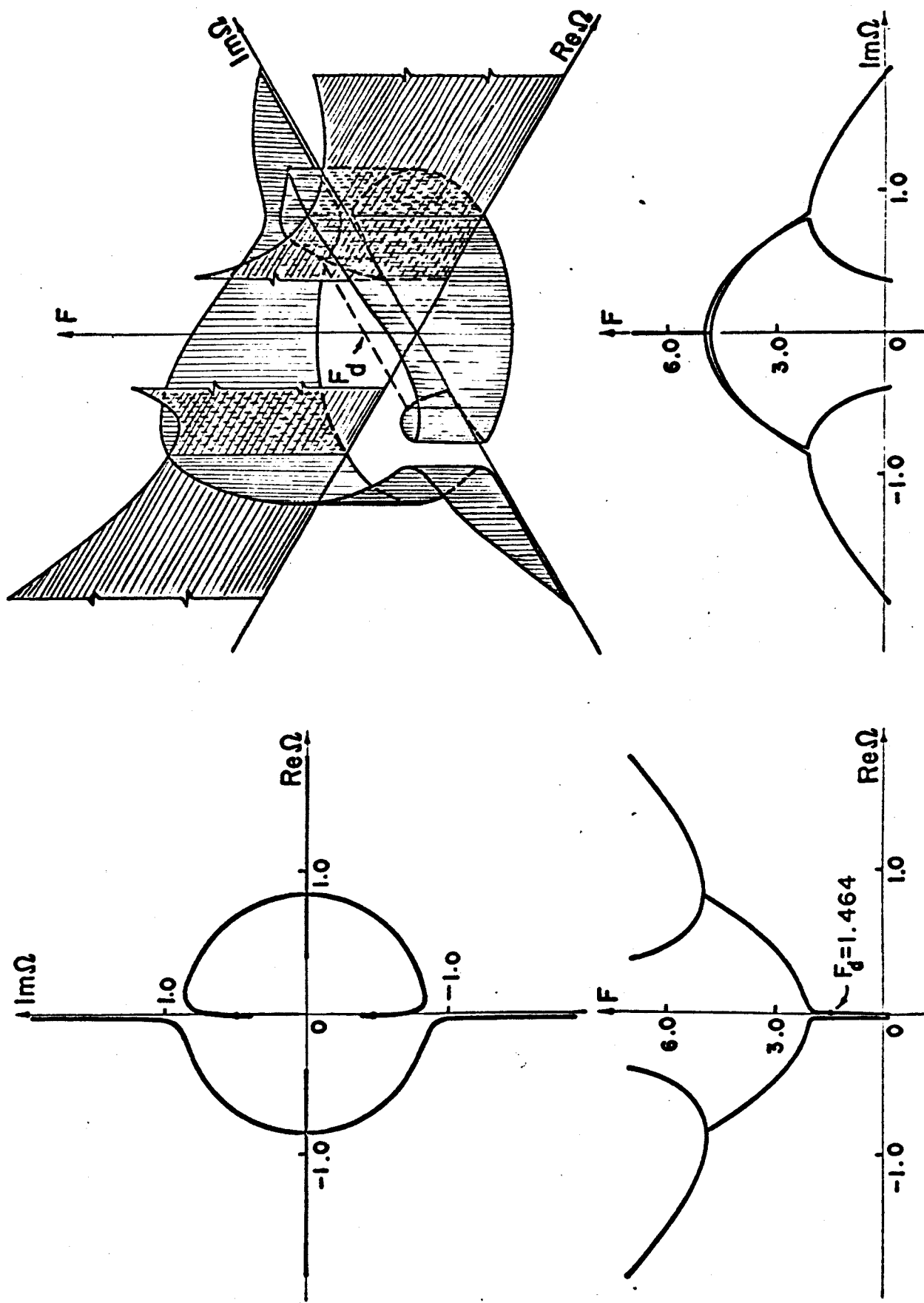


Fig. 3

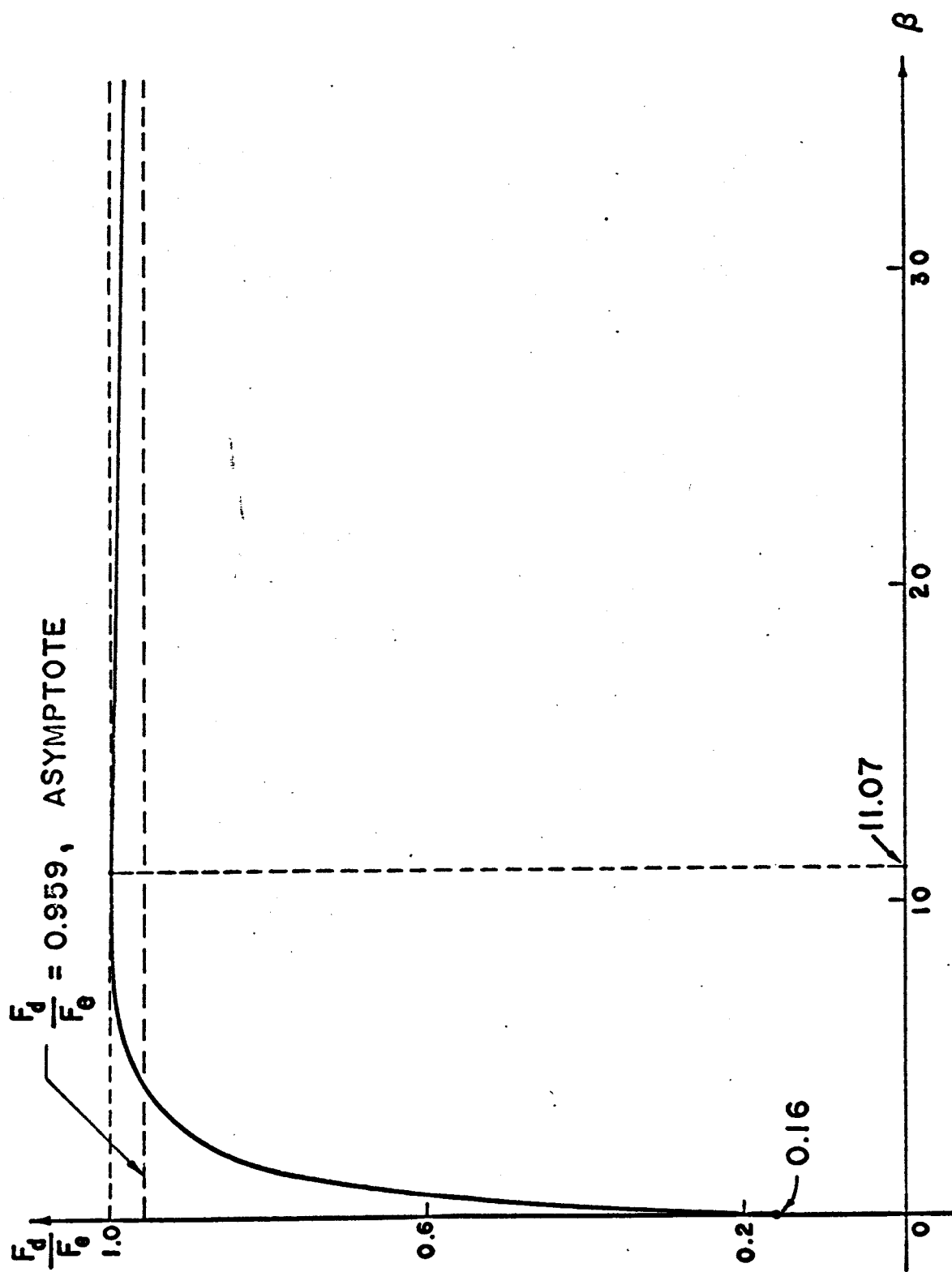


Fig. 4

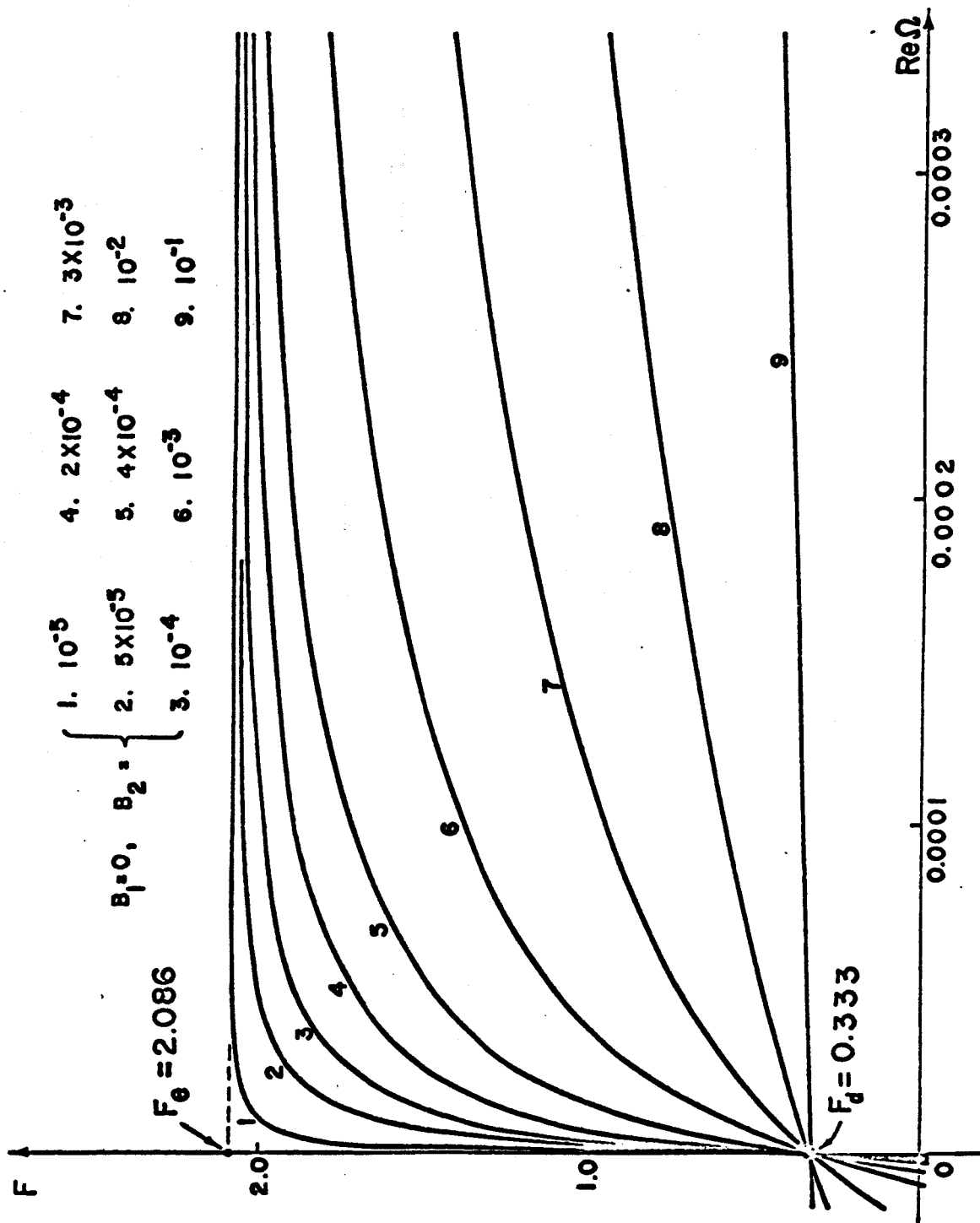


Fig. 5

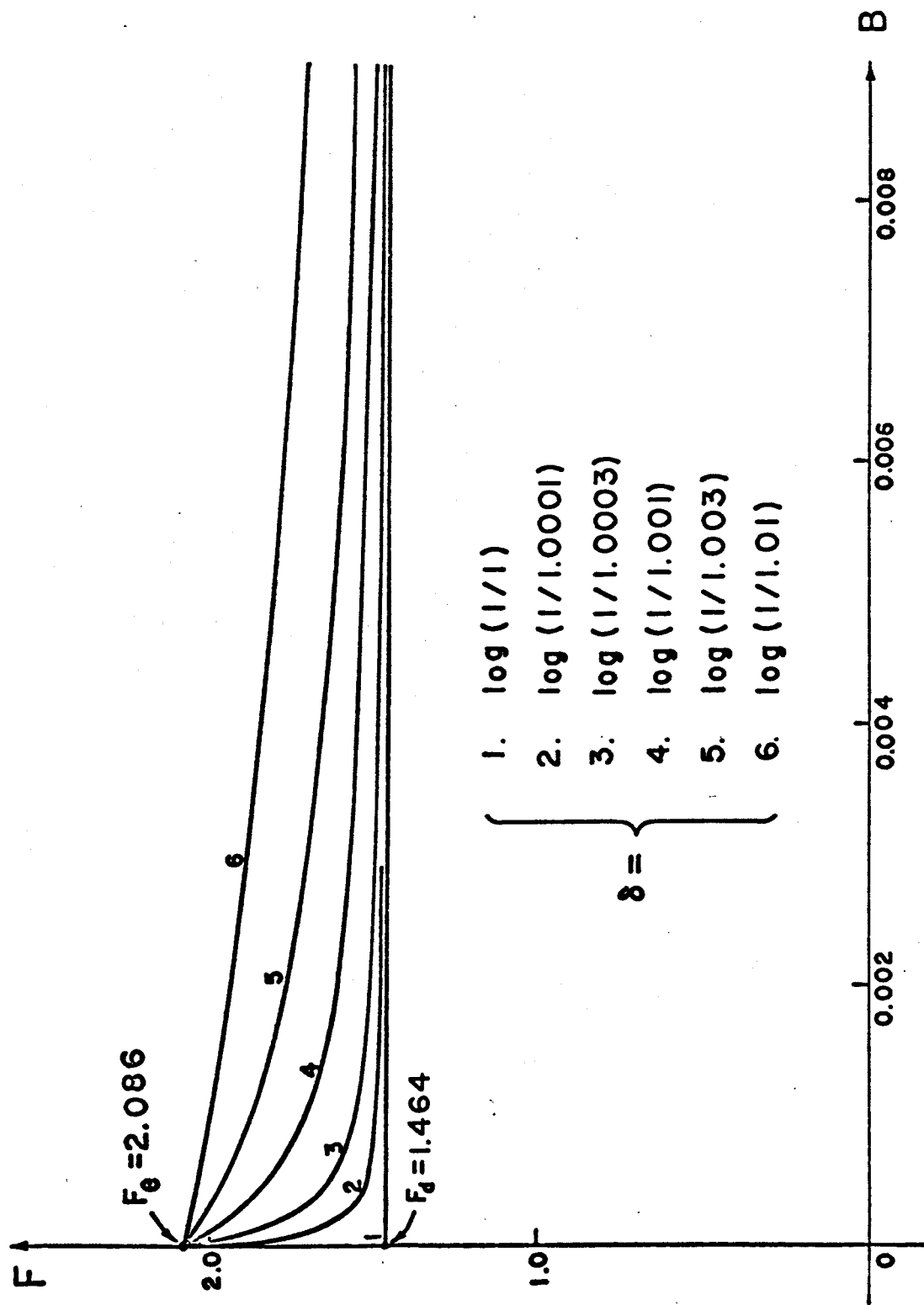


Fig. 6

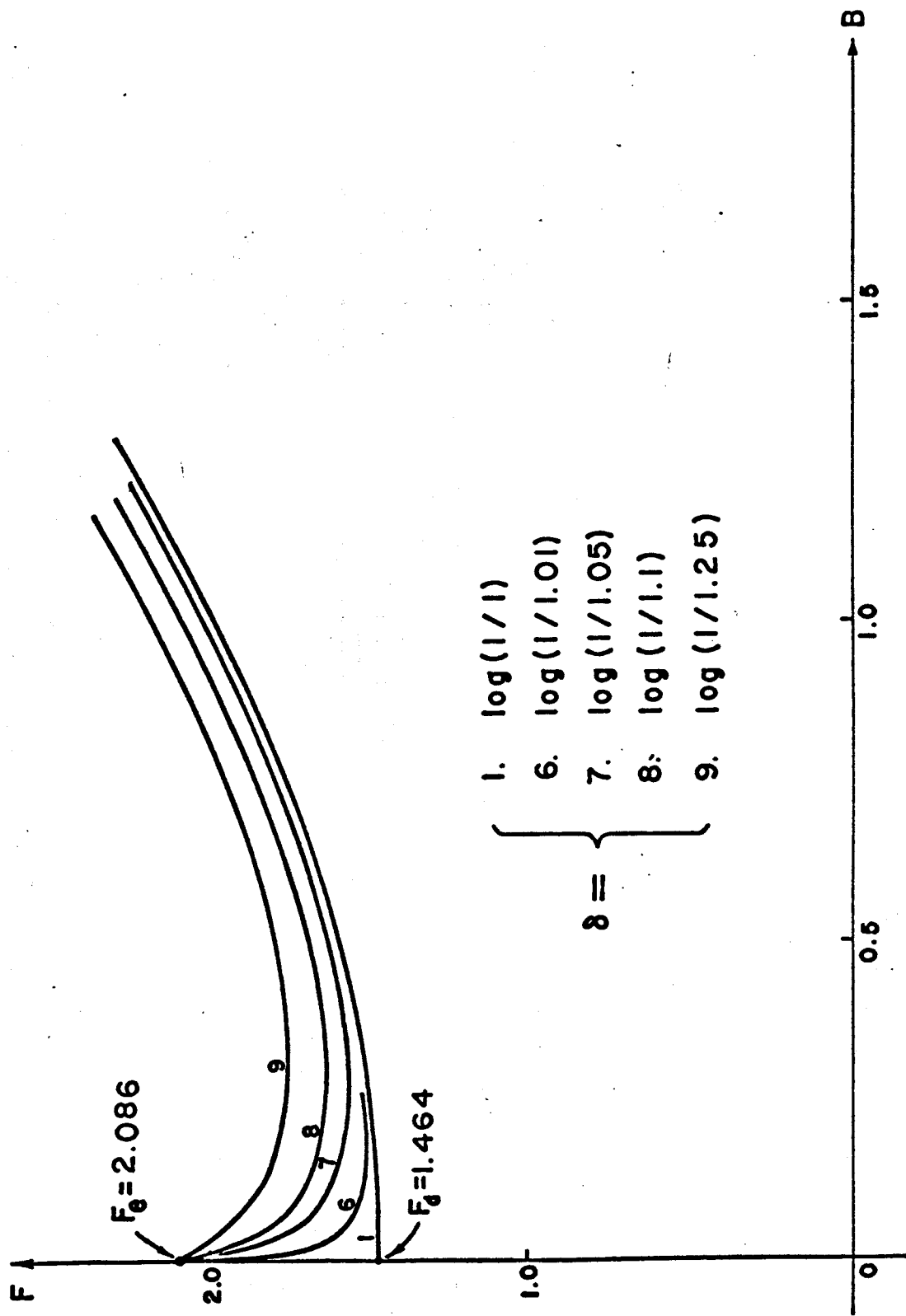


Fig. 7