

T.&A.M. REPORT NO. 281

# A METHOD FOR DETERMINING AN OPTIMUM SHAPE OF A CLASS OF THIN SHELLS OF REVOLUTION

GPO PRICE \$ \_\_\_\_\_

CSFTI PRICE(S) \$ \_\_\_\_\_

Hard copy (HC) 3.00

Microfiche (MF) 50

by

Morris Stern  
Han-chung Wang  
Will J. Worley

ff 653 July 65

*NSG-734*

Prepared Under Grant No. (NGR-14-005-010)  
NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
WASHINGTON, D.C.

**N 65-33503**

FACILITY FORM 602

(ACCESSION NUMBER)

(THRU)

(PAGE)

(CODE)

(NUMBER OF TEXT OR FIGURE)

(CATEGORY)

DEPARTMENT OF THEORETICAL AND APPLIED MECHANICS  
UNIVERSITY OF ILLINOIS  
URBANA, ILLINOIS

A METHOD FOR DETERMINING AN OPTIMUM SHAPE OF  
A CLASS OF THIN SHELLS OF REVOLUTION

by

Morris Stern  
Han-chung Wang  
Will J. Worley

Prepared under Grant No. NGR 14-005-010

by the

Department of Theoretical and Applied Mechanics  
UNIVERSITY OF ILLINOIS  
URBANA, ILLINOIS

for

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
WASHINGTON, D. C.  
July, 1965

## TABLE OF CONTENTS

	Page
SUMMARY - - - - -	1
INTRODUCTION - - - - -	2
1. Statement of the Problem - - - - -	2
2. Symbols - - - - -	4
3. Acknowledgment - - - - -	6
DISCUSSION OF THE METHOD - - - - -	7
MATHEMATICAL FORMULATION - - - - -	10
NUMERICAL INTEGRATION - - - - -	16
NUMERICAL COMPUTATIONS - - - - -	20
1. The Ranges of the Volumes and Weights of Shells - - - - -	20
2. Weighting Functions $\lambda$ and $\mu$ - - - - -	22
3. Numerical Examples - - - - -	23
APPENDIXES - - - - -	27
A. Iteration Procedure with Varying Shell Length - - - - -	27
B. FORTRAN Programs - - - - -	28
REFERENCES - - - - -	39
TABLES - - - - -	40
I. Iteration Results for $\lambda = \lambda_1, \mu = \mu_1, b/a = 1, \alpha_0 = 1.5, \beta_0 = 1.5$ -	40
II. Iteration Results for $\lambda = \lambda_1, \mu = \mu_1, b/a = 1, \alpha_0 = 2.0, \beta_0 = 3.0$ -	41
III. Iteration Results for $\lambda = \lambda_1, \mu = \mu_1, b/a = 2, \alpha_0 = 1.5, \beta_0 = 1.5$ -	42
IV. Iteration Results for $\lambda = \lambda_1, \mu = \mu_1, b/a = 2, \alpha_0 = 2.0, \beta_0 = 3.0$ -	43
V. Iteration Results for $\lambda = \lambda_2, \mu = \mu_2, b/a = 1$ - - - - -	44
VI. Iteration Results for $\lambda = \lambda_3, \mu = \mu_3, b/a = 1$ - - - - -	45
FIGURES - - - - -	46
1. Weighting Functions $\lambda_1$ and $\mu_1$ - - - - -	46
2. Weighting Functions $\lambda_2$ and $\mu_2$ - - - - -	46
3. Variation of the Function F with $\alpha$ and $\beta$ - - - - -	47
4. Variation in Projections of Ridges on $\alpha - \beta$ Plane - - - - -	48
5. Variation in $V_{xa}/ab^2$ with $\alpha$ and $\beta$ - - - - -	49
6. Variation in $A/a^2$ with $\alpha$ and $\beta$ for $b/a = 1.0$ - - - - -	50

A METHOD FOR DETERMINING AN OPTIMUM SHAPE OF  
A CLASS OF THIN SHELLS OF REVOLUTION

by

Morris Stern, Han-chung Wang, Will J. Worley

Department of Theoretical and Applied Mechanics  
University of Illinois  
Urbana, Illinois

SUMMARY

33503

This third report under the current grant is concerned with a method for determining an optimum shape of a convex shell of revolution with respect to volume, weight and length.

The technique used depends on replacing the class of functions, over which the shape may range, by the parameters  $b/a$ ,  $\alpha$  and  $\beta$  in the equation

$$\left| \frac{x}{a} \right|^{\alpha} + \left| \frac{y}{b} \right|^{\beta} = 1$$

where  $a$ ,  $b$ ,  $\alpha$  and  $\beta$  are positive constants not necessarily integers, with  $\alpha$  and  $\beta$  equal to or greater than unity. The bodies of revolution are generated by revolving the line, described by the above equation, about the x-axis.

The procedure is illustrated for a thin shell which will fit within the space defined by a circular cylinder of radius  $b$  and length  $2a$ . The shell is optimized, in terms of  $\alpha$  and  $\beta$ , with respect to volume and weight. The FORTRAN program used to achieve these results is presented in Appendix B.

*Author*

## INTRODUCTION

### 1. Statement of the Problem.

The previous reports under the current grant, [1,2] \* stated a future objective of the project as being the optimum contour design of a class of shells. This third report is directed toward achieving that objective in terms of enclosed volume and shell weight for thin shells of revolution.

Optimization can be treated in several ways. A general formulation of the optimization of the design of thin shells of revolution might include the determination of the shell shape as well as the variation of the shell thickness along meridional lines. A less general approach involves assigning the shape and varying the shell thickness [3, 4]. The current report treats an alternate approach. Here a uniform thickness is maintained, but the meridional lines which define the geometry are permitted to vary in accordance with the relation

$$\left| \frac{x}{a} \right|^{\alpha} + \left| \frac{y}{b} \right|^{\beta} = 1 \quad (3.1)**$$

where  $a$ ,  $b$ ,  $\alpha$  and  $\beta$  are positive constants, not necessarily integers.

The use of Eq. (3.1) permits an optimization of shape which is limited to the choice of the parameters  $\alpha$  and  $\beta$  for a shell of length  $2a$  and of radius  $b$ . The body of this report is limited to the variation of  $\alpha$  and  $\beta$  for fixed length and fixed diameter, but Appendix A presents a mathematical formulation which permits the length to vary as well as  $\alpha$  and  $\beta$ .

The achievement of the stated objective depends on a suitable failure criterion. One criterion could involve a complete stress analysis of the shell including varying thickness. Others could include thick walled shells or buckling. However, in illustrating the method, the shells have been restricted to thin, constant thickness walls with internal pressure loading. Further the failure is assumed to occur either on the central plane circle normal to the  $x$ -axis at  $x = 0$  or along a meridian. Thus separate computer programs which involve the complete stress analysis of the shell have not been used.

---

\*Numbers in brackets refer to the References.

\*\*The notation (3.1) is adopted to aid in cross-referencing equations from the first two reports under the grant [1, 2].

The techniques described can be applied in a manner which would permit the direct inclusion of one of the existing computer programs on the stress analysis of shells [5, 6, 7] . These auxilliary computer programs would provide the thickness requirement or the variation in thickness of the shell when incorporated into the proper location within the FORTRAN program presented in this report. In this way the optimized shell would be based on a more realistic failure criterion than is actually reported.

## 2. Symbols

a	half length of the shell, [L]*
b	radius of the shell in the equatorial plane, [L]
x	horizontal coordinate of the first quadrant of Eq. (3.1), [L]
y	vertical coordinate of the first quadrant of Eq. (3.1), [L]
g	acceleration due to gravity, [LT <sup>-2</sup> ]
V <sub>xa</sub>	volume of the shell, [L <sup>3</sup> ]
W	weight of the shell, [MLT <sup>-2</sup> ]
A	surface area of the shell, [L <sup>2</sup> ]
A <sub>a</sub>	area enclosed by first quadrant of Eq. (3.1), [L <sup>2</sup> ]
L	arc length in the first quadrant of Eq. (3.1), [L]
t	thickness of the shell, [L]
V <sub>min</sub>	preassigned minimum allowable volume of the shell, [L <sup>3</sup> ]
W <sub>max</sub>	preassigned maximum allowable weight of the shell, [MLT <sup>-2</sup> ]
a <sub>max</sub>	preassigned maximum half length of the shell, [L]
V <sub>cyl</sub>	volume of cylinder with radius b, length 2a, [L <sup>3</sup> ]
W <sub>cyl</sub>	weight of cylindrical shell with radius b, length 2a, [MLT <sup>-2</sup> ]
v	ratio of V <sub>xa</sub> /V <sub>min</sub> , [1]
w	ratio of W/W <sub>max</sub> , [1]
ℓ	ratio of a/a <sub>max</sub> , [1]
h <sup>2</sup>	$\left(\frac{b\alpha}{a\beta}\right)^2$ , [1]

---

\*The dimensional notation [L] indicates a length while [M] indicates mass, [T] indicates time and [1] indicates a dimensionless quantity.

$p_0$	uniform internal pressure on shells, $[ML^{-1} T^{-2}]$
$k_0$	preselected limiting value for the ratio $\Delta\alpha/F_\alpha$ or $\Delta\beta/F_\beta$ of iteration, [1]
$\alpha$	exponent of the absolute value of $x/a$ , [1]
$\beta$	exponent of the absolute value of $y/b$ , [1]
$\alpha$	(as a subscript) indicates partial differentiation with respect to $\alpha$ , [1]
$\beta$	(as a subscript) indicates partial differentiation with respect to $\beta$ , [1]
$\rho$	mass density, $[ML^{-3}]$
$\lambda$	non-negative weighting function of $v$ , [1]
$\mu$	non-negative weighting function of $w$ , [1]
$\nu$	non-negative weighting function of $\ell$ , [1]
$\sigma_0$	yield stress of the shell material, $[ML^{-1} T^{-2}]$
$\eta_0^2$	preselected limiting value for the maximum change in $(\Delta v^2 + \Delta w^2)$ to be allowed in one iteration step, [1]
$J_1$ through $J_7$	integrals as defined in Eqs. (3.33)
$I(\epsilon)$ , $K(\epsilon)$	improper integrals as defined in Eqs. (3.35) and (3.36)



### 3. Acknowledgment

This project was sponsored by the National Aeronautics and Space Administration, Office of Advanced Research and Technology, Applied Mathematics Branch, of which Dr. Raymond H. Wilson is Chief.

The investigation was part of the work of the Engineering Experiment Station of which Professor Ross J. Martin is Director and was conducted in the Department of Theoretical and Applied Mechanics of which Professor Thomas J. Dolan is Head, with Will J. Worley as Principal Investigator.

The authors wish to acknowledge the assistance of Charles Cecil Fretwell, formerly Instructor in Theoretical and Applied Mechanics, University of Illinois, in the early stages of the numerical programming and the assistance of undergraduate students: Messrs. Tom E. Breuer and Edward H. Stredde with various phases of the project.

The suggestion that the length variation be included as a parameter in the optimization procedure was made by Melvin G. Rosche, Space Vehicle Structures Program, NASA, Washington, D. C.

Both the ILLIAC II and the IBM 7094 computer facilities were used. The ILLIAC II was constructed in the Digital Computer Laboratory, now known as the Department of Computer Science, University of Illinois with support from the Atomic Energy Commission, grant USAEC AT(11-1)-415, and from the Office of Naval Research, grant NONOR-1832 (15). The IBM 7094 computer facility is partially supported by the National Science Foundation under grant NSF GP700.

## DISCUSSION OF THE METHOD

Otpimization with respect to enclosed volume and shell weight, for a shell of revolution defined by the meridian curve Eq. (3.1) is achieved by considering the exponents  $\alpha$  and  $\beta$  as parameters.

Then the volume and weight may be expressed as

$$V_{xa} = V_{xa}(\alpha, \beta) \quad (3.2)$$

$$W = \rho g A(\alpha, \beta) t(\alpha, \beta) \quad (3.3)$$

where  $\rho$  is the mass density of the material of the shell,  $g$  is the gravitational acceleration, while  $A$  represents the area of the middle surface of the shell and  $t$  is the thickness. The thickness is maintained constant over the entire shell and is small compared to the radius  $b$  and to the length  $a$ .

Both the volume  $V_{xa}$  and the surface area  $A$  depend only on the geometrical shape of the shell, which is controlled by the parameters  $\alpha$  and  $\beta$  for the fixed cylindrical volume. The thickness  $t$  depends on the geometrical shape of the shell, on the load condition and on the failure criteria. Therefore the mode of the failure of the shell, under a specified load condition, must be defined for the evaluation of the thickness  $t$ , before optimization can be achieved.

Let the primary design requirements, to be fulfilled for the shell, be

$$V_{xa} \geq V_{\min} \quad , \quad W \leq W_{\max}$$

where  $V_{\min}$  and  $W_{\max}$  are preassigned limits. It is further assumed that at least one set of values  $(\alpha, \beta)$  will satisfy the primary requirements. Otherwise the material of the shell, the assumed mode of failure, the load conditions, or the dimensions  $a$  and  $b$  have to be modified in order to determine an optimum shape.

To facilitate the calculations, the ratios of the volume and weight are introduced as

$$v = \frac{V_{xa}}{V_{\min}} \quad , \quad w = \frac{W}{W_{\max}} \quad (3.4)$$

The differential of a function is then defined as

$$dF = \lambda dv - \mu dw \quad (3.5)$$

where  $\lambda$  and  $\mu$  are non-negative weighting functions of  $v$  and  $w$ , which define the relative importance of increasing in volume and of decreasing in weight. These weighting functions are defined in terms of current volume and weight. As long as it is possible to select  $dv$  and  $dw$ , consistent with the constraints of the problem, such that  $dF$  is positive, one has not achieved the optimum shape. Thus one seeks the values of  $\alpha$  and  $\beta$  for which the differential  $dF$  is either zero or negative.

While superior shapes may exist, the above criteria will assure an optimum shape within the limitations of Eq. (3.1) and with the imposed constraints on volume and weight.

To determine the values of  $\alpha$  and  $\beta$  for which  $F$  yields the extreme value, one may write Eq. (3.5) in the form

$$dF = F_{\alpha} d\alpha + F_{\beta} d\beta \quad (3.6)$$

with

$$F_{\alpha} = \lambda v_{\alpha} - \mu w_{\alpha} \quad (3.7)$$

$$F_{\beta} = \lambda v_{\beta} - \mu w_{\beta} \quad (3.8)$$

where the subscripts  $\alpha$  and  $\beta$  indicate the partial differentiation with respect to  $\alpha$  and  $\beta$ .

If Eq. (3.6) is an exact differential, then in principle one need only look among the solutions of  $F_{\alpha} = F_{\beta} = 0$  for the optimum shape. Because of the complex nature of the equations for  $v_{\alpha}$ ,  $v_{\beta}$ ,  $w_{\alpha}$  and  $w_{\beta}$ , it is difficult to determine whether Eq. (3.6) is exact. Even if Eq. (3.6) were exact, the analytical solution of  $F_{\alpha} = F_{\beta} = 0$  would be extremely difficult to obtain. The following iterative procedure is therefore used in the evaluation of  $F_{\alpha} = F_{\beta} = 0$ .

A shape defined by a set of  $\alpha$  and  $\beta$  consistent with the primary requirements is selected first. This defines the shape of the shell middle surface. Therefore, the volume and the surface area of the shell can be calculated and the required thickness computed consistent with the assumed mode failure of the shell. Once the volume and weight are computed, values of  $\lambda$  and  $\mu$ , which were defined by the design criterion, are established. Hence the values of  $F_{\alpha}$  and  $F_{\beta}$  are determined by Eqs. (3.7) and (3.8). The shape is then modified by incrementing  $\alpha$  and  $\beta$  in accordance with the path of the steepest ascent

$$d\alpha : d\beta = F_{\alpha} : F_{\beta} \quad (3.9)$$

The iterative procedure is repeated until  $F_\alpha$  and  $F_\beta$  are both essentially zero.

To determine the incremental size  $\Delta\alpha$  and  $\Delta\beta$  for the steps in the iteration, let a constant  $k$  be defined from Eq. (3.9) as

$$\frac{\Delta\alpha}{F_\alpha} = \frac{\Delta\beta}{F_\beta} = k \quad (3.10)$$

Therefore

$$dv = v_\alpha d\alpha + v_\beta d\beta = k(v_\alpha F_\alpha + v_\beta F_\beta) \quad (3.11)$$

$$dw = w_\alpha d\alpha + w_\beta d\beta = k(w_\alpha F_\alpha + w_\beta F_\beta) \quad (3.12)$$

In order to limit the size of the increments of  $\Delta v$  and  $\Delta w$ , and of  $\Delta\alpha$  and  $\Delta\beta$ , the constant  $k$  is selected in the following way

$$k = \begin{cases} k_1 & \text{if } k_1 < k_0 \\ k_0 & \text{if } k_1 > k_0 \end{cases} \quad (3.13)$$

The constant  $k_1$  is determined from Eqs. (3.11) and (3.12) consistent with the assigned increments of  $\Delta v$  and  $\Delta w$ , and is evaluated as follows

$$\eta_0^2 = \Delta v^2 + \Delta w^2 = k_1^2 \left[ (v_\alpha F_\alpha + v_\beta F_\beta)^2 + (w_\alpha F_\alpha + w_\beta F_\beta)^2 \right]$$

from which

$$k_1 = \eta_0 / \left[ (v_\alpha F_\alpha + v_\beta F_\beta)^2 + (w_\alpha F_\alpha + w_\beta F_\beta)^2 \right]^{1/2} \quad (3.14)$$

where  $\eta_0^2$  is a preselected limiting value for the maximum change of  $(\Delta v^2 + \Delta w^2)$  to be allowed in one iteration step, while  $k_0$  is a preselected limiting value for the ratio  $\Delta\alpha/F_\alpha$  or  $\Delta\beta/F_\beta$  for each step of iteration. The process is then repeated with a new set of values of  $\alpha$  and  $\beta$  formed by adding the increments  $\Delta\alpha$  and  $\Delta\beta$  to the previous values. The iteration process terminates when the value  $(F_\alpha^2 + F_\beta^2)$  is less than a preassigned accuracy parameter.

The mathematical formulation of the more general problem which permits the length to vary as well as  $\alpha$  and  $\beta$  is presented in Appendix A.

## MATHEMATICAL FORMULATION

In the process of iteration, as described in the previous sections, the values of  $v$ ,  $v_\alpha$ ,  $v_\beta$ ,  $w$ ,  $w_\alpha$  and  $w_\beta$  for a given set of values of  $\alpha$  and  $\beta$  must be calculated. From Eqs. (3.3) and (3.4),  $w_\alpha$  and  $w_\beta$  may be written as

$$w_\alpha = \frac{\rho g}{W_{\max}} (A t_\alpha + A_\alpha t) \quad (3.15)$$

$$w_\beta = \frac{\rho g}{W_{\max}} (A t_\beta + A_\beta t) \quad (3.16)$$

The symbols used in the iteration procedure, described earlier in the report, are defined by the following integrals. The notation in these integrals is consistent with that used in the previous reports under the current research grant [1, 2].

$$\begin{aligned} v &= \frac{2 \pi b^2}{V_{\min}} \int_0^a \left[ 1 - \left( \frac{x}{a} \right)^\alpha \right]^{2/\beta} dx \\ &= \frac{2 \pi ab^2}{V_{\min}} \int_0^1 (1 - X^\alpha)^{2/\beta} dX \end{aligned} \quad (3.17)$$

$$v_\alpha = \frac{-2 \pi ab^2}{V_{\min}} \left( \frac{2}{\beta} \right) \int_0^1 (1 - X^\alpha)^{(2-\beta)/\beta} X^\alpha \log X dX \quad (3.18)$$

$$v_\beta = \frac{-2 \pi ab^2}{V_{\min}} \left( \frac{2}{\beta^2} \right) \int_0^1 (1 - X^\alpha)^{2/\beta} \log(1 - X^\alpha) dX \quad (3.19)$$

$$\begin{aligned} A &= 4 \pi b \int_0^a \left[ 1 - \left( \frac{x}{a} \right)^\alpha \right]^{1/\beta} \left\{ 1 + \left( \frac{b\alpha}{a\beta} \right)^2 \left( \frac{x}{a} \right)^{2(\alpha-1)} \left[ 1 - \left( \frac{x}{a} \right)^\alpha \right]^{2(1-\beta)/\beta} \right\}^{1/2} dx \\ &= 4 \pi ab \int_0^1 (1 - X^\alpha)^{1/\beta} \left[ 1 + \left( \frac{b\alpha}{a\beta} \right)^2 X^{2(\alpha-1)} (1 - X^\alpha)^{2(1-\beta)/\beta} \right]^{1/2} dX \end{aligned} \quad (3.20)$$

Let  $F(X, \alpha, \beta) = 1 + \left(\frac{b\alpha}{a\beta}\right)^2 X^{2(\alpha-1)} (1 - X^\alpha)^{2(1-\beta)/\beta}$

and  $h^2 = \left(\frac{b\alpha}{a\beta}\right)^2$

then  $A_\alpha = 4 \pi ab \left\{ \left(\frac{h^2}{\alpha}\right) \int_0^1 F(X, \alpha, \beta)^{-1/2} X^{2(\alpha-1)} (1 - X^\alpha)^{(3-2\beta)/\beta} dX \right.$   
 $- \frac{1}{\beta} \int_0^1 F(X, \alpha, \beta)^{-1/2} X^\alpha (1 - X^\alpha)^{(1-\beta)/\beta} \log X dX$   
 $+ h^2 \int_0^1 F(X, \alpha, \beta)^{-1/2} X^{2(\alpha-1)} (1 - X^\alpha)^{(3-2\beta)/\beta} \log X dX$   
 $\left. - h^2 \left(\frac{2-\beta}{\beta}\right) \int_0^1 F(X, \alpha, \beta)^{-1/2} X^{(3\alpha-2)} (1 - X^\alpha)^{3(1-\beta)/\beta} \log X dX \right\} \quad (3.21)$

and

$$A_\beta = -4 \pi ab \left\{ \left(\frac{h^2}{\beta}\right) \int_0^1 F(X, \alpha, \beta)^{-1/2} X^{2(\alpha-1)} (1 - X^\alpha)^{(3-2\beta)/\beta} dX \right.$$

$$+ \frac{1}{\beta^2} \int_0^1 F(X, \alpha, \beta)^{-1/2} (1 - X^\alpha)^{1/\beta} \log (1 - X^\alpha) dX$$

$$\left. + 2 \left(\frac{h^2}{\beta^2}\right) \int_0^1 F(X, \alpha, \beta)^{-1/2} X^{2(\alpha-1)} (1 - X^\alpha)^{(3-2\beta)/\beta} \log (1 - X^\alpha) dX \right\} \quad (3.22)$$

The next step consists of the determination of the thickness  $t$  and the values of  $t_\alpha$  and  $t_\beta$ . These values should ideally be determined from a limit analysis, but since this would constitute a major undertaking in itself [6, 7], the following simple failure criterion is adopted. It is assumed that under a uniform internal pressure,  $p_0$ , the shell will fail by general yielding either along a longitudinal plane or around

the equatorial plane. If  $\sigma_0$  is the yield stress for the shell material, failure along a longitudinal plane requires a thickness given by

$$t_1 = \left( \frac{p_0}{\sigma_0} \right) \frac{A_a}{L} \quad (3.23)$$

while failure around the equatorial plane requires a thickness given by

$$t_2 = \frac{1}{2} \left( \frac{p_0}{\sigma_0} \right) b \quad (3.24)$$

where  $A_a$  is the area enclosed by the first quadrant of Eq. (3.1) and  $L$  is the complete arc length in the first quadrant of Eq. (3.1). The design thickness  $t$  should be either  $t_1$  or  $t_2$ , whichever is larger. If  $t_2 \geq t_1$  then  $t$  equals  $t_2$ , a constant; therefore  $t_\alpha = t_\beta = 0$ . For  $t_2 < t_1$ , then by Eq. (3.23),

$$t_\alpha = \frac{p_0}{\sigma_0} \left[ \frac{(A_a)_\alpha}{L} - \frac{A_a L_\alpha}{L^2} \right] \quad (3.25)$$

$$t_\beta = \frac{p_0}{\sigma_0} \left[ \frac{(A_a)_\beta}{L} - \frac{A_a L_\beta}{L^2} \right] \quad (3.26)$$

where

$$A_a = b \int_0^a \left[ 1 - \left( \frac{x}{a} \right)^\alpha \right]^{1/\beta} dx = ab \int_0^1 (1 - X^\alpha)^{1/\beta} dX \quad (3.27)$$

$$(A_a)_\alpha = \frac{\partial A_a}{\partial \alpha} = - \left( \frac{ab}{\beta} \right) \int_0^1 (1 - X^\alpha)^{(1-\beta)/\beta} X^\alpha \log X dX \quad (3.28)$$

$$(A_a)_\beta = \frac{\partial A_a}{\partial \beta} = - \left( \frac{ab}{\beta^2} \right) \int_0^1 (1 - X^\alpha)^{1/\beta} \log(1 - X^\alpha) dX \quad (3.29)$$

$$\begin{aligned}
L &= \int_0^a \left\{ 1 + \left( \frac{b\alpha}{a\beta} \right)^2 \left( \frac{x}{a} \right)^{2(\alpha-1)} \left[ 1 - \left( \frac{x}{a} \right)^\alpha \right]^{2(1-\beta)/\beta} \right\}^{1/2} dx \\
&= a \int_0^1 F(X, \alpha, \beta)^{1/2} dX
\end{aligned} \tag{3.30}$$

$$\begin{aligned}
L_\alpha &= a \left\{ \frac{h^2}{\alpha} \int_0^1 F(X, \alpha, \beta)^{-1/2} X^{2(\alpha-1)} (1-X^\alpha)^{2(1-\beta)/\beta} dX \right. \\
&\quad + h^2 \int_0^1 F(X, \alpha, \beta)^{-1/2} X^{2(\alpha-1)} (1-X^\alpha)^{(2-3\beta)/\beta} \log X dX \\
&\quad \left. - \frac{h^2}{\beta} \int_0^1 F(X, \alpha, \beta)^{-1/2} X^{3\alpha-2} (1-X^\alpha)^{(2-3\beta)/\beta} \log X dX \right\}
\end{aligned} \tag{3.31}$$

$$\begin{aligned}
L_\beta &= a \left\{ -\frac{h^2}{\beta} \int_0^1 F(X, \alpha, \beta)^{-1/2} X^{2(\alpha-1)} (1-X^\alpha)^{2(1-\beta)/\beta} dX \right. \\
&\quad \left. - \frac{h^2}{\beta^2} \int_0^1 F(X, \alpha, \beta)^{-1/2} X^{2(\alpha-1)} (1-X^\alpha)^{2(1-\beta)/\beta} \log(1-X^\alpha) dX \right\}
\end{aligned} \tag{3.32}$$

All of the integrals which appear in the above equations may be collected into seven groups, by defining the following convergent but sometimes improper integrals in notations as

$$J_1(p, q) = \int_0^1 (1 - u^p)^q du$$

$$J_2(p, q) = \int_0^1 (1 - u^p)^q \log(1 - u^p) du$$



$$J_3(p, q) = \int_0^1 (1 - u^p)^{q-1} u^p \log u \, du$$

$$J_4(p, q, s) = \int_0^1 (1 - u^p)^s \left[ 1 + h^2 u^{2(p-1)} (1 - u^p)^{2(q-1)} \right]^{1/2} du$$

(3.33)

$$J_5(p, q, s) = \int_0^1 \left[ 1 + h^2 u^{2(p-1)} (1 - u^p)^{2(q-1)} \right]^{-1/2} u^{2(p-1)} (1 - u^p)^{s-2} du$$

$$J_6(p, q, r, s) = \int_0^1 \left[ 1 + h^2 u^{2(p-1)} (1 - u^p)^{2(q-1)} \right]^{-1/2} u^{2(r-1)} (1 - u^p)^{s-2} \log(1 - u^p) du$$

$$J_7(p, q, s, m) = \int_0^1 \left[ 1 + h^2 u^{2(p-1)} (1 - u^p)^{2(q-1)} \right]^{-1/2} u^m (1 - u^p)^{s-3} \log u \, du$$

with  $h^2 = \left( \frac{b\alpha}{a\beta} \right)^2$  and  $p, r \geq 1, q, s, m \geq 0$ .

Then the equations used in the calculation of  $v, w$  and their derivatives can be expressed by Eqs. (3.33) as

$$v = \frac{2\pi ab^2}{V_{\min}} J_1\left(\alpha, \frac{2}{\beta}\right)$$

$$v_\alpha = -\frac{2\pi ab^2}{V_{\min}} \left(\frac{2}{\beta}\right) J_3\left(\alpha, \frac{2}{\beta}\right)$$

$$v_\beta = -\frac{2\pi ab^2}{V_{\min}} \left(\frac{2}{\beta^2}\right) J_2\left(\alpha, \frac{2}{\beta}\right)$$

$$A = 4 \pi ab J_4\left(\alpha, \frac{1}{\beta}, \frac{1}{\beta}\right)$$

$$A_\alpha = 4 \pi ab \left[ \frac{h^2}{\alpha} J_5\left(\alpha, \frac{1}{\beta}, \frac{3}{\beta}\right) - \frac{1}{\beta} J_7\left(\alpha, \frac{1}{\beta}, \frac{1}{\beta} + 2, \alpha\right) \right. \\ \left. + h^2 J_7\left(\alpha, \frac{1}{\beta}, \frac{3}{\beta} + 1, 2\alpha - 2\right) - h^2 \left(\frac{2}{\beta} - 1\right) J_7\left(\alpha, \frac{1}{\beta}, \frac{3}{\beta}, 3\alpha - 2\right) \right]$$

$$A_\beta = -4 \pi ab \left[ \frac{h^2}{\beta} J_5\left(\alpha, \frac{1}{\beta}, \frac{3}{\beta}\right) + \frac{1}{\beta^2} J_6\left(\alpha, \frac{1}{\beta}, 1, \frac{1}{\beta} + 2\right) \right. \\ \left. + 2 \left(\frac{h}{\beta}\right)^2 J_6\left(\alpha, \frac{1}{\beta}, \alpha, \frac{3}{\beta}\right) \right]$$

(3.34)

$$A_a = ab J_1\left(\alpha, \frac{1}{\beta}\right)$$

$$(A_a)_\alpha = ab \left(-\frac{1}{\beta}\right) J_3\left(\alpha, \frac{1}{\beta}\right)$$

$$(A_a)_\beta = ab \left(-\frac{1}{\beta^2}\right) J_2\left(\alpha, \frac{1}{\beta}\right)$$

$$L = a J_4\left(\alpha, \frac{1}{\beta}, 0\right)$$

$$L_\alpha = a \left[ \frac{h^2}{\alpha} J_5\left(\alpha, \frac{1}{\beta}, \frac{2}{\beta}\right) + h^2 J_7\left(\alpha, \frac{1}{\beta}, \frac{2}{\beta}, 2\alpha - 2\right) \right. \\ \left. - \frac{h^2}{\beta} J_7\left(\alpha, \frac{1}{\beta}, \frac{2}{\beta}, 3\alpha - 2\right) \right]$$

$$L_\beta = -a \left[ \frac{h^2}{\beta} J_5\left(\alpha, \frac{1}{\beta}, \frac{2}{\beta}\right) + \frac{h^2}{\beta^2} J_6\left(\alpha, \frac{1}{\beta}, \alpha, \frac{2}{\beta}\right) \right]$$

## NUMERICAL INTEGRATION

The analytical expressions for the integrals in Eqs.(3.33) are not available except for  $J_1$ ,  $J_2$  and  $J_3$  which may be expressed in terms of gamma functions and the derivatives of gamma functions, the psi functions. Since the above functions also involve series expansions or numerical integration, all of the integrals in Eqs.(3.33) are evaluated numerically using Simpson's Rule. In this process, special consideration is given to improper integrals of the following two types.

$$I(\epsilon) = \int_0^{\epsilon} f(\xi) \xi^{\delta} d\xi \quad \delta > -1 \quad (3.35)$$

$$K(\epsilon) = \int_0^{\epsilon} f(\xi) \xi^{\delta} \log \xi d\xi \quad \delta > -1 \quad (3.36)$$

with  $f(\xi)$  continuous in the interval  $0 \leq \xi \leq \epsilon$ . For  $\epsilon$  small enough, replace  $f(\xi)$  with a parabola

$$f(\xi) = b_0 + b_1 \left(\frac{\xi}{\epsilon}\right) + b_2 \left(\frac{\xi}{\epsilon}\right)^2 \quad (3.37)$$

where

$$\begin{aligned} b_0 &= f(0) \\ b_1 &= 4 f\left(\frac{\epsilon}{2}\right) - f(\epsilon) - 3 f(0) \\ b_2 &= 2 f(\epsilon) - 4 f\left(\frac{\epsilon}{2}\right) + 2 f(0) \end{aligned}$$

Then the improper integrals  $I(\epsilon)$  and  $K(\epsilon)$  can be approximated as

$$\begin{aligned} I(\epsilon) &= \int_0^{\epsilon} \left[ b_0 + b_1 \left(\frac{\xi}{\epsilon}\right) + b_2 \left(\frac{\xi}{\epsilon}\right)^2 \right] \xi^{\delta} d\xi \\ &= \frac{\epsilon^{\delta+1}}{\delta+1} \left[ b_0 + \frac{\delta+1}{\delta+2} b_1 + \frac{\delta+1}{\delta+3} b_2 \right] \end{aligned} \quad (3.38)$$

$$\begin{aligned}
K(\epsilon) &= \int_0^\epsilon \left[ b_0 + b_1 \left( \frac{\xi}{\epsilon} \right) + b_2 \left( \frac{\xi}{\epsilon} \right)^2 \right] \xi^\delta \log \xi \, d\xi \\
&= \frac{\epsilon^{\delta+1}}{\delta+1} \left\{ \log \epsilon \left[ b_0 + \frac{\delta+1}{\delta+2} b_1 + \frac{\delta+1}{\delta+3} b_2 \right] - \frac{1}{\delta+1} \left[ b_0 + \left( \frac{\delta+1}{\delta+2} \right)^2 b_1 + \left( \frac{\delta+1}{\delta+3} \right)^2 b_2 \right] \right\} \quad (3.39)
\end{aligned}$$

Therefore the improper integrals  $J_2$  to  $J_7$  listed in Eqs.(3.33) can be expressed in terms of  $I(\epsilon)$ ,  $K(\epsilon)$  and a proper integral. They are derived as follows.

$$J_2(p, q) = \int_0^{1-\eta} (1-u^p)^q \log(1-u^p) \, du + \int_{1-\eta}^1 (1-u^p)^q \log(1-u^p) \, du$$

Let  $\xi = 1 - u^p$

then  $du = -\frac{1}{p} (1-\xi)^{(1-p)/p} \, d\xi$

and  $J_2(p, q) = \int_0^{1-\eta} (1-u^p)^q \log(1-u^p) \, du + \frac{1}{p} \int_0^{1-(1-\eta)^p} (1-\xi)^{(1-p)/p} \xi^q \log \xi \, d\xi$

Using Eq. (3.39) along with the definitions of Eq. (3.37),  $J_2$  is approximated as

$$J_2(p, q) = \int_0^{1-\eta} (1-u^p)^q \log(1-u^p) \, du + \frac{1}{p} K(\epsilon) \quad \left| \begin{array}{l} * \\ \epsilon = 1-(1-\eta)^p \\ f(\xi) = (1-\xi)^{(1-p)/p} \\ \delta = q \end{array} \right. \quad (3.40)$$

In a similar manner, by the approximations of  $I(\epsilon)$  and  $K(\epsilon)$ , other integrals yield;

$$\left. \begin{aligned}
J_3(p, q) &= \int_\epsilon^{1-\eta} (1-u^p)^{q-1} u^p \log u \, du \\
&+ \frac{1}{p^2} I(\epsilon) \quad \left| \begin{array}{l} \epsilon = 1-(1-\eta)^p \\ f(\xi) = (1-\xi)^{1/p} \log(1-\xi) \\ \delta = q-1 \end{array} \right. \quad + K(\epsilon) \quad \left| \begin{array}{l} \epsilon = \epsilon \\ f(\xi) = (1-\xi^p)^{q-1} \\ \delta = p \end{array} \right.
\end{aligned} \right\}$$

\*This notation indicates that the integral is evaluated at the indicated values of  $\epsilon$ ,  $f(\xi)$  and  $\delta$ .

$$J_4(p, q, s) = \int_0^{1-\eta} (1-u^p)^s [1+h^2 u^{2(p-1)} (1-u^p)^{2(q-1)}]^{1/2} du$$

$$+\frac{h}{p} I(\epsilon) \left| \begin{array}{l} \epsilon = 1-(1-\eta)^p \\ f(\xi) = \left[ \frac{1}{h^2} (1-\xi)^{2(1-p)/p} \xi^{2(1-q)} + 1 \right]^{1/2} \\ \delta = q + s - 1 \end{array} \right.$$

$$J_5(p, q, s) = \int_0^{1-\eta} [1+h^2 u^{2(p-1)} (1-u^p)^{2(q-1)}]^{-1/2} u^{2(p-1)} (1-u^p)^{s-2} du$$

(3.41)

$$+\frac{1}{p} I(\epsilon) \left| \begin{array}{l} \epsilon = 1-(1-\eta)^p \\ f(\xi) = \left[ \xi^{2(1-q)} + h^2 (1-\xi)^{2(p-1)/p} \right]^{-1/2} (1-\xi)^{(p-1)/p} \\ \delta = s - q - 1 \end{array} \right.$$

$$J_6(p, q, r, s) = \int_0^{1-\eta} [1+h^2 u^{2(p-1)} (1-u^p)^{2(q-1)}]^{-1/2} u^{2(r-1)} (1-u^p)^{s-2}$$

$$\log(1-u^p) du$$

$$+\frac{1}{p} K(\epsilon) \left| \begin{array}{l} \epsilon = 1-(1-\eta)^p \\ f(\xi) = \left[ \xi^{2(1-q)} + h^2 (1-\xi)^{2(p-1)/p} \right]^{-1/2} (1-\xi)^{(2r-p-1)/p} \\ \delta = s - q - 1 \end{array} \right.$$

$$\begin{aligned}
J_7(p, q, s, m) = & \int_{\epsilon}^{1-\eta} \left[ 1 + h^2 u^{2(p-1)} (1-u)^{2(q-1)} \right]^{-1/2} (1-u)^{s-3} u^m \log u \, du \\
& + K(\epsilon) \left| \begin{array}{l} \epsilon = \epsilon \\ f(\xi) = \left[ (1-\xi)^{2(1-q)} + h^2 \xi^{2(p-1)} \right]^{-1/2} (1-\xi)^{s-q-2} \\ \delta = m \end{array} \right. \\
& + \frac{1}{p^2} I(\epsilon) \left| \begin{array}{l} \epsilon = 1 - (1-\eta)^p \\ f(\xi) = \left[ \xi^{2(1-q)} + h^2 (1-\xi)^{2(p-1)/p} \right]^{-1/2} (1-\xi)^{(m+1-p)/p} \\ \quad \frac{1}{\xi} \log(1-\xi) \\ \delta = s - q - 1 \end{array} \right.
\end{aligned}$$

The integrals  $J_1$  to  $J_7$  therefore involve only proper integrals; thus numerical integration by Simpson's rule can be applied. The FORTRAN programs for the evaluation of  $J_1$  through  $J_7$  by means of a digital computer are written in subfunction form as listed in Appendix B.

## NUMERICAL COMPUTATIONS

Several characteristics of the design problem have to be defined before the numerical iterations can be performed. One is the magnitude of the required values for  $V_{\min}$  and  $W_{\max}$ , the others include the weighting functions  $\lambda$  and  $\mu$ .

### 1. The Ranges of the Volumes and Weights of Shells

Among the shells of revolution which may be generated by revolving the meridian curve, Eq. (3.1), about the x-axis, the range of shapes of interest lie between the cylindrical shell for which the exponents  $\alpha$  and  $\beta$  are both large, and the conical shell for which the exponents  $\alpha$  and  $\beta$  both equal unity. The volume of the cylinder with radius  $b$  and length  $2a$  is  $V_{\text{cyl}} = 2\pi ab^2$ , and the volume of the double cone, having apexes at  $-a$  and at  $+a$ , with corresponding base diameter,  $2b$ , is  $V_{\text{cone}} = \frac{2}{3}\pi ab^2$ . If the required minimum volume  $V_{\min}$  is written as  $V_{\min} = C_1(2\pi ab^2)$ , then  $C_1$  must lie between  $1/3$  and  $1$ .

In order to determine the weight for the two limiting cases, the thickness variation must be considered as well as the surface area. The surface areas for the cylindrical and conical shells are

$$A_{\text{cyl}} = 4\pi ab \left[ 1 + \frac{1}{2} \left( \frac{b}{a} \right) \right]$$

$$A_{\text{cone}} = 4\pi ab \left[ \frac{1}{2} \sqrt{1 + \left( \frac{b}{a} \right)^2} \right]$$

Using Eqs. (3.23) and (3.24), the thicknesses required for the shell of cylindrical type, based on two different failure criteria, are expressed as

$$t_1 = \frac{1}{1 + \left( \frac{b}{a} \right)} \left( \frac{p_0}{\sigma_0} \right) b$$

$$t_2 = \frac{1}{2} \left( \frac{p_0}{\sigma_0} \right) b$$

therefore

$$t = \left\{ \begin{array}{l} \frac{1}{2} \left( \frac{p_0}{\sigma_0} \right) b \quad \text{for } \left( \frac{b}{a} \right) \geq 1 \\ \frac{1}{1 + \left( \frac{b}{a} \right)} \left( \frac{p_0}{\sigma_0} \right) b \quad \text{for } \left( \frac{b}{a} \right) < 1 \end{array} \right\}$$

Similarly, the thicknesses required for the conical type shell are

$$t_1 = \frac{1}{2} \frac{1}{\sqrt{1 + \left(\frac{b}{a}\right)^2}} \left(\frac{p_0}{\sigma_0}\right) b$$

$$t_2 = \frac{1}{2} \left(\frac{p_0}{\sigma_0}\right) b$$

Since  $\frac{1}{\sqrt{1 + \left(\frac{b}{a}\right)^2}} \leq 1$  for any value of  $\left(\frac{b}{a}\right)$ , it follows that

$$t = \frac{1}{2} \left(\frac{p_0}{\sigma_0}\right) b \quad \text{for any ratio } \left(\frac{b}{a}\right).$$

Then the weights of the cylindrical and the conical shells become

$$W_{\text{cyl}} = \left\{ \begin{array}{l} \frac{1}{2} \left[1 + \frac{1}{2} \left(\frac{b}{a}\right)\right] \left[4 \pi ab^2 \left(\frac{p_0}{\sigma_0}\right) \rho g\right] \quad \text{for } \frac{b}{a} \geq 1 \\ \frac{1}{1 + \left(\frac{b}{a}\right)} \left[1 + \frac{1}{2} \left(\frac{b}{a}\right)\right] \left[4 \pi ab^2 \left(\frac{p_0}{\sigma_0}\right) \rho g\right] \quad \text{for } \frac{b}{a} < 1 \end{array} \right\} \quad (3.42)$$

$$W_{\text{cone}} = \frac{1}{4} \sqrt{1 + \left(\frac{b}{a}\right)^2} \left[4 \pi ab^2 \left(\frac{p_0}{\sigma_0}\right) \rho g\right]$$

By writing the primary required limiting weight as

$$W_{\text{max}} = C_2 \left[4 \pi ab^2 \left(\frac{p_0}{\sigma_0}\right) \rho g\right]$$

then  $C_2$  has to be in the range

$$\frac{1}{4} \sqrt{1 + \left(\frac{b}{a}\right)^2} \leq C_2 \leq \frac{1}{2} \left[1 + \frac{1}{2} \left(\frac{b}{a}\right)\right] \quad \text{for } \left(\frac{b}{a}\right) \geq 1$$

$$\frac{1}{4} \sqrt{1 + \left(\frac{b}{a}\right)^2} \leq C_2 \leq \frac{1}{1 + \left(\frac{b}{a}\right)} \left[1 + \frac{1}{2} \left(\frac{b}{a}\right)\right] \quad \text{for } \left(\frac{b}{a}\right) < 1$$



## 2. Weighting Functions $\lambda$ and $\mu$

The functions  $\lambda$  and  $\mu$  which define the relative importance of the variation of the volume and the weight of the shells, are preassigned according to the design criterion. Any functions in terms of  $V$  and  $W$  can be assigned in the problem. One such set of functions is defined as

$$\lambda_1 = \frac{(V_{\text{cyl}} - V)^p}{(V - V_{\text{min}})^q}, \quad \mu_1 = \frac{W^m}{(W_{\text{max}} - W)^n} \quad (3.43)$$

The shapes of the functions in Eq. (3.43) appear in Fig. 1, for  $p = q = m = n = 1$ . From the characteristics of the functions  $\lambda$  and  $\mu$ , one can predict that when the volume is close to  $V_{\text{min}}$  or when the weight is close to  $W_{\text{max}}$ , a small increment of  $V$  or  $W$  will produce a large change of  $dF$  as defined in Eq. (3.6). If  $dF$  is considered as the slope of a surface  $F$ , then the surface has a positive slope along the edge where  $V$  is close to  $V_{\text{min}}$  and has a negative slope along the edge where  $W$  is close to  $W_{\text{max}}$ . Thus it follows that there must exist a maximum value of  $F$ , that is  $dF = 0$ , in the assigned range  $V > V_{\text{min}}$  and  $W < W_{\text{max}}$ .

The functions  $\lambda$  and  $\mu$  may also be defined as

$$\lambda_2 = \frac{\lambda_1}{\sqrt{\lambda_1^2 + \mu_1^2}}, \quad \mu_2 = \frac{\mu_1}{\sqrt{\lambda_1^2 + \mu_1^2}} \quad (3.44)$$

where  $\lambda_1$  and  $\mu_1$  are obtained from Eq. (3.43). If one divides  $\lambda_1$  and  $\mu_1$  by  $\sqrt{\lambda_1^2 + \mu_1^2}$ , the magnitude of  $\lambda_2$  and  $\mu_2$  will be limited to the range of 0 to 1. The ratios  $\lambda_2/\mu_2$  and  $\lambda_1/\mu_1$  remain the same. The characteristics of the functions are as shown in Fig. 2; the values of  $\lambda_2$  and  $\mu_2$  outside the range of  $V_{\text{min}}$  and  $W_{\text{max}}$  are arbitrary set equal to 0 and to 1 respectively.

Another form of  $\lambda$  and  $\mu$  consists of straight lines, which define a linear variation of  $V$  and  $W$  as

$$\lambda_3 = \frac{V_{\text{cyl}} - V}{V_{\text{cyl}} - V_{\text{min}}}, \quad \mu_3 = \frac{W}{W_{\text{max}}} \quad (3.45)$$

The above three definitions of  $\lambda$  and  $\mu$  are applied in the numerical examples of this report. Subroutine programs for the calculation of  $\lambda$  and  $\mu$  are attached to the main iteration program as listed in Appendix B.

### 3. Numerical Examples

The functions of  $\lambda$  and  $\mu$  for the first example are chosen as in Fig. 1, that is

$$\lambda = \frac{V_{\text{cyl}} - V}{V - V_{\text{min}}}, \quad \mu = \frac{W}{W_{\text{max}} - W}$$

Let the required minimum volume of the shell be 0.6 of the volume of cylindrical shell, thus  $V_{\text{min}} = 0.6 (2 \pi a b^2)$ , and let the required maximum weight be 0.8 that of the cylindrical shell under the same load condition. Since the thickness required for the cylindrical shell is dependent on the ratio of  $b/a$ , the weights for the cylindrical shells with different ratios  $b/a$  are given by Eq. (3.42) as

$$W_{\text{cyl}} = \begin{cases} 0.75 \\ 1.00 \\ 0.90 \end{cases} (4 \pi \rho g p_0 a b^2 / \sigma_0) \begin{matrix} \text{for } b/a = 1.0 \\ \text{for } b/a = 2.0 \\ \text{for } b/a = 0.25 \end{matrix}$$

therefore the values of  $W_{\text{max}}$  are chosen as  $0.6 (4 \pi \rho g p_0 a b^2 / \sigma_0)$ ,  $0.8 (4 \pi \rho g p_0 a b^2 / \sigma_0)$  and  $0.72 (4 \pi \rho g p_0 a b^2 / \sigma_0)$  for the ratios  $b/a = 1.0, 2.0,$  and  $0.25$  respectively.

With all the requirements set, the iterative calculations are performed with the aid of digital computers. The FORTRAN programs for the iteration procedure are listed in Appendix B.

Choosing the starting values  $\alpha = 1.5, \beta = 1.5$  with the fineness ratio  $a/b = 1.0$ , and the limiting value  $\eta_0 = 0.1, k_0 = 0.3$ , results in the output listed in Table I. From steps 1 to 7 in the Table, the results listed are presented for each iteration; from step 8 on, the results are presented for every other iteration. From these results it is seen that the values of  $\alpha$  and  $\beta$  increase rapidly in each of the first six iterations and then change slowly. The same pattern is apparent for the slopes  $F_\alpha$  and  $F_\beta$ .

Table II shows the iteration results with the same parameters as in Table I, but with the starting condition  $\alpha = 2.0, \beta = 3.0$ . The results in steps 1 to 8 are listed for each iteration while after step 8, they are listed for every fourth iteration. The iterated values of  $\alpha$  and  $\beta$  decrease rapidly in the first three steps and then change slowly.

The results in Tables I and II, indicate that the shape defined by  $\alpha = 1.5, \beta = 1.5$  lies on one side of the ridge, Fig. 3, while the shape defined by  $\alpha = 2.0, \beta = 3.0$  lies on the other side of the ridge. During the iteration process, the successively improving values of  $\alpha$  and  $\beta$  climb to the ridge rapidly according to the path of steepest ascent, and then progress slowly along the ridge due to the small variation of slope along the ridge.

The phenomena observed in the above results may be verified or described more clearly by the exact integration of the function  $dF$  of Eq. (3.6) using the assigned functions  $\lambda$  and  $\mu$ . Rewriting the weighting functions  $\lambda$  and  $\mu$  in dimensionless terms  $v$  and  $w$ , one obtains

$$\lambda = \frac{c - v}{v - 1} \qquad \mu = \frac{w}{1 - w}$$

where 
$$c = \frac{V_{\text{cyl}}}{V_{\text{min}}}$$

Then 
$$dF = \frac{c - v}{v - 1} dv - \frac{w}{1 - w} dw$$

which, after integration, yields

$$F = (c - 1) \log(v - 1) + \log(1 - w) - (v + w)$$

The function  $F$  is in terms of  $v$  and  $w$ , which can be represented by the integrals with parameters  $\alpha$  and  $\beta$ . The relative variation of  $F$  with respect to  $\alpha$  and  $\beta$  is plotted as a three dimensional surface in Fig. 3. The surface has the shape of a mountain range with the projection of the ridge shown in the  $\alpha - \beta$  plane in Fig. 3. The peak of the ridge is located near the point  $\alpha = 2.65$ ,  $\beta = 1.55$ .

Changing the values of  $V_{\text{min}}$ ,  $W_{\text{max}}$  and the reciprocal of fineness ratio,  $b/a$ , results in little change in the shape of the surface  $F$ , but does produce a slight shift in the location of the ridge. The projections of the ridges on the  $\alpha - \beta$  plane, with different combinations of  $V_{\text{min}}$ ,  $W_{\text{max}}$ , and  $b/a$ , are plotted in Fig. 4. The shift in the ridge is in the same sense as the change in  $V_{\text{min}}$  or in  $W_{\text{max}}$ .

The results in Tables III and IV show the iterative calculations for the case  $V_{\text{min}} = 0.6 (2 \pi ab^2)$ ,  $W_{\text{max}} = 0.8 (4 \pi \rho g p_0 ab^2 / \sigma_0)$  with the ratio  $b/a = 2.0$  and  $k_0 = 0.3$ . While the initial values for  $\alpha$  and  $\beta$  are different in Tables III and IV, it is noted that they converge to the same values of  $\alpha$  and  $\beta$  after successive iterations.

As a second example,  $\lambda_2$  and  $\mu_2$  of Fig. 2 are chosen as the weighting functions. The results of each iterated calculation with three different starting values are listed in Table V. The preassigned values for computations are  $V_{\text{min}} = 0.6 (2 \pi ab^2)$ ,  $W_{\text{max}} = 0.6 (4 \pi \rho g p_0 ab^2 / \sigma_0)$  and  $b/a = 1.0$ . The values

of  $\alpha$  and  $\beta$  reach the ridge rapidly after several iterations regardless of the starting point.

Table VI gives the iterated results for the weighting functions  $\lambda_3$  and  $\mu_3$ , which vary linearly with  $V$  and  $W$ , as defined by Eq. (3.45). The other preassigned values for computations are the same as for Table V. The results indicate that both the slopes  $F_\alpha$  and  $F_\beta$  are within the limit 0.005 after ten iterations for all of the different starting values.

Since the iteration procedure is controlled by the slope of the function  $F$ , the rate of convergence is mainly dependent on the weighting functions  $\lambda$  and  $\mu$ . For the currently assigned functions, the results in the above tables indicate that the values  $\alpha$  and  $\beta$  converge rapidly to the region where the ordered pair  $(\alpha, \beta)$  lies near the projection of the ridge and then change slowly along the ridge. Due to the small variation of the slope along the ridge, any point located on the projection of the ridge on the  $\alpha - \beta$  plane constitutes a good shape with respect to volume and weight.

As another example, the functions  $\lambda$  and  $\mu$  may be considered as constants. In this case, the problem becomes one of determining the relative maximum of the function  $F = v - w$ . Since the variation of the thickness is very small due to the change of values  $\alpha, \beta$ , a shape which is nearly optimum may be achieved by assigning a specific value of volume in determining the values of  $\alpha, \beta$  for minimum shell surface or by assigning a specific value of surface area in determining the values of  $\alpha, \beta$  for maximum volume.

The shapes to fulfill the above requirement can be determined with the aid of data from previous reports [1, 2]. The surface in Fig. 5 represents the volume variation with respect to  $\alpha$  and  $\beta$ . The heavy curve on this surface represents the volumes of shells for which the surface area is equal to a preassigned value. From the projection of this curve on a vertical plane, the values of  $\alpha$  and  $\beta$  for the maximum volume for the defined surface area can be established. In a similar manner, the surface in Fig. 6 represents the area variation with respect to  $\alpha$  and  $\beta$ . The heavy curve on the surface represents the areas of shells for which the shell volume is equal to a preassigned value. The projection of this curve on a vertical plane indicates the area variation among shells having a constant volume.

## APPENDIXES

### A. Iteration Procedures with Varying Shell Length

Similar to the Eqs. (3.2) and (3.3), the volume and weight of the shells of revolution may be taken as the functions of three parameters  $\alpha$ ,  $\beta$  and  $a$ ,

$$V_{xa} = V_{xa}(\alpha, \beta, a)$$

$$W = \rho g A(\alpha, \beta, a) t(\alpha, \beta, a)$$

Here the shape requirements to be fulfilled for the shell are

$$V_{xa} \geq V_{\min} \quad W \leq W_{\max} \quad a \leq a_{\max}$$

In this case the dimensionless forms are defined as

$$v = \frac{V_{xa}}{V_{\min}} \quad w = \frac{W}{W_{\max}} \quad \ell = \frac{a}{a_{\max}}$$

and the differential of a function is formed as

$$dF = \lambda dv - \mu dw - \nu d\ell$$

where  $\lambda$ ,  $\mu$  and  $\nu$  are the functions defining the relative importance of increases in volume and decreases in weight and length. The differential  $dF$  also can be written as

$$dF = F_{\alpha} d\alpha + F_{\beta} d\beta + F_{\ell} d\ell$$

with

$$F_{\alpha} = \lambda v_{\alpha} - \mu w_{\alpha}$$

$$F_{\beta} = \lambda v_{\beta} - \mu w_{\beta}$$

$$F_{\ell} = \lambda v_{\ell} - \mu w_{\ell} - \nu$$

The iteration steps will then follow path of steepest ascent, as defined by

$$d\alpha : d\beta : d\ell = F_{\alpha} : F_{\beta} : F_{\ell}$$

The iterative procedure is repeated until  $F_{\alpha}$ ,  $F_{\beta}$  and  $F_{\ell}$  are essentially zero.

## B. FORTRAN Programs

### 1. Programs for FUNCTIONS FJ1 through FJ7

Since the integrals  $J_1$  through  $J_7$  listed in Eqs. (3.33) appear in the calculations of the main iteration program many times, they are computed in separate FUNCTIONS attached to the main program. Simpson's rule is used to evaluate the above integrals with the approximation techniques discussed in the section on Numerical Integration.

Among the input arguments for the FUNCTION programs, the values P, Q, R, S and T are the exponents in the integrals. They are dependent on the values of  $\alpha$  and  $\beta$ . The quantities ETAI and EPS are two small numbers assigned in the calculation of the two improper integrals  $I(\epsilon)$  and  $K(\epsilon)$  of Eqs. (3.35) and (3.36). The value FK2 represents the term  $\left(\frac{b\alpha}{a\beta}\right)^2$  and the value ACC is the accuracy required for the relative difference between two successive approximations in the Simpson's rule integration routine. In the previous numerical examples, the value assigned to ETAI and to EPS is 0.01 while the value assigned to ACC is 0.0001.

### 2. Program for SUBROUTINE FMULAM

The SUBROUTINE FMULAM is written to compute either the values  $\lambda_1$ ,  $\mu_1$ , as Eq. (3.43) or the values  $\lambda_2$ ,  $\mu_2$  as Eq. (3.44), which is controlled by the number NC. The outputs defined by FLAM and FMU represent  $\lambda_1$  and  $\mu_1$  for  $NC = 1$  and  $\lambda_2$  and  $\mu_2$  for  $NC = 2$ . The input arguments P, Q, FM and FN are the same as the exponents p, q, m and n of Eq. (3.43).

### 3. The Main Iteration Program

The main purpose of the program is to compute the increments of  $\Delta\alpha$  and  $\Delta\beta$  along the path of the steepest ascent from the current assigned values  $\alpha$  and  $\beta$ . The computations are repeated for the new calculated  $\alpha$  and  $\beta$  until they reach a point where the absolute values  $F_\alpha$  and  $F_\beta$ , as in Eqs. (3.7) and (3.8), are less than a preassigned small number QEPS.

The input data of FP, FQ, FM, FN and NC listed on the first data card are supplied for the calculation of functions  $\lambda$  and  $\mu$ . The constants EPS, ETAI and ACC on the second data card are the numbers assigned to the FUNCTIONS  $J_1$  through  $J_7$  in order to compute the integrals. The values PO and SIGO represent internal pressure  $p_0$  and the yield stress  $\sigma_0$  and are used to calculate the thickness t. Input data ETAO and FKV are assigned to limit the step size of  $\alpha$  and

$\beta$  in each iteration, and represent  $\eta_0$  and  $k_0$  in Eqs. (3.14) and (3.13). The two integers NRVWP and NRAB are the number of the sets of  $V_{\min}$ ,  $W_{\max}$  and the number of sets of the ratio BOA ( $b/a$ ) to be calculated in the program.

The input values VMIN and WMAX are two dimensionless numbers which represent the preassigned allowable minimum volume and maximum weight. The true value of the minimum volume is  $VMIN \cdot (2 \pi ab^2)$  and the true value of maximum weight is  $WMAX \cdot (4 \pi \rho g p_0 ab^2/\sigma_0)$ .

For the output, the results of each iteration are printed using the symbols DA, DB, ATIL and BTIL to represent  $\Delta\alpha$ ,  $\Delta\beta$ ,  $F_\alpha$  and  $F_\beta$  respectively.

## PROGRAM FOR EXAMPLES

```

$   FORTRAN IBM
$   PUNCH OBJECT
$   GO
   READ INPUT TAPE 7, 1, FP, FQ, FM, FN, NC, EPS, ETAI, ACC, NRWVP SHELL001
1  FORMAT (4F15.5, I10 / 3E20.8 / I10) SHELL002
   READ INPUT TAPE 7, 2, PU, SIGG, QEPS, ETAD, FKV SHELL003
2  FORMAT (2E20.8/ 3F20.5) SHELL004
3  DO 69 INRVWP = 1, NRWVP SHELL005
4  READ INPUT TAPE 7, 5, VMIN, WMAX, NRAB SHELL006
5  FORMAT ( 2E20.8 / I10) SHELL007
9  DO 50 INRAB = 1, NRAB SHELL008
10 READ INPUT TAPE 7, 11, ALPHA, BETA, BOA SHELL009
11 FORMAT ( 3E20.8) SHELL010
   WRITE OUTPUT TAPE 6, 12, VMIN, WMAX, BOA SHELL011
12 FORMAT (8H1VMIN = ,F7.3,2X,7HWMAX = ,F7.3,3X, 4HPQA=, F6.3, //) SHELL012
   NCONT = 0 SHELL013
C SHELL014
C   TO LIMIT ALPHA AND BETA BOTH LARGER THAN ONE SHELL015
C SHELL016
14 IF (ALPHA - 1.0) 15, 17, 17 SHELL017
15 ALPHA = 1.0 SHELL018
17 IF (BETA - 1.0) 18, 19, 19 SHELL019
18 BETA = 1.0 SHELL020
19 FK2 = (BOA*ALPHA/BETA)**2 SHELL021
   CNOB = 1.0/BETA SHELL022
   TWOB = 2.0/BETA SHELL023
   THOB = 3.0/BETA SHELL024
   NCONT = NCONT +1 SHELL025
C SHELL026
C   TO DETERMINE THE VALUE OF T= T1 OR T2 SHELL027
C SHELL028
   AR= FJ1(ALPHA,CNOB,ACC) SHELL029
   SL = FJ4(ALPHA,CNOB,0.0,FK2,ETAI,ACC) SHELL030
   RR = AR/ SL SHELL031
   IF ( RR-0.5 ) 20, 20, 21 SHELL032
C SHELL033
C   T2 LARGER THAN T1 SHELL034
C SHELL035
20 T = 0.5 *PO/SIGO SHELL036
   TA = C. SHELL037
   TB = C. SHELL038
   GO TO 24 SHELL039
C SHELL040
C   T1 LARGER THAN T2 SHELL041
C SHELL042
21 T = RR *PC /SIGO SHELL043
   ARA= -(CNOB)*FJ3(ALPHA,CNOB,CPS,ETAI,ACC) SHELL044
   ARB= -(1.0/(BETA**2))*FJ2(ALPHA,CNOB,ETAI,ACC) SHELL045
   TE = FJ5(ALPHA,CNOB,TWOB,FK2,ETAI,ACC) SHELL046
   SLA = FK2*(FJ7(ALPHA,CNOB,TWOB,2.*(ALPHA-1.),FK2,EPS,ETAI,ACC) - SHELL047
1   FJ7(ALPHA,CNOB,TWOB,3.*ALPHA-2.,FK2,EPS,ETAI,ACC)/BETA + SHELL048
2   TE/ALPHA) SHELL049
   SLB = -(FK2/(BETA**2))*(FJ6(ALPHA,CNOB,ALPHA,TWOB,FK2,ETAI,ACC) + SHELL050
1   BETA*TE) SHELL051
   TA = (PC/SIGO) * ( ARA- AR*SLA/SL ) / SL SHELL052
   TB = (PU/SIGO) * ( ARB- AR * SLB/ SL ) /SL SHELL053
24 WRITE OUTPUT TAPE 6, 25, ALPHA, BETA, AR, SL, RR, T, TA, TB
25 FORMAT ( 1HC, 5X, 8F12.6)

```



C		SHELL054
C	TO CALCULATE V, W, AND THEIR DERIVATIVES	SHELL055
C		SHELL056
	SU= FJ4(ALPHA,ONOB,ONOB,FK2,ETAI,ACC)	SHELL057
	TE = FK2*FJ5(ALPHA,ONOB,THOB,FK2,ETAI,ACC)	SHELL058
	SUA= TE/ALPHA - FJ7(ALPHA,ONOB, ONOB+2., ALPHA,FK2,EPS,ETAI,ACC)	SHELL059
	1 /BETA + FK2*FJ7(ALPHA,ONOB, THOB+1.0, 2.*(ALPHA-1.),FK2,EPS,	SHELL060
	2 ETAI,ACC) - FK2*(TWOB-1.)*FJ7(ALPHA,ONOB,THOB,3.*ALPHA-2.,	SHELL061
	3 FK2,EPS,ETAI,ACC)	SHELL062
	SUB= -(1./(BETA**2))*(FJ6(ALPHA,ONOB,1.,ONOB+2.,FK2,ETAI,ACC) +	SHELL063
	1 2.*FK2*FJ6(ALPHA,ONOB,ALPHA,THOB,FK2,ETAI,ACC) + BETA*TE)	SHELL064
	V = FJ1(ALPHA,TWOB,ACC)	SHELL065
	VA = -(TWOB)*FJ3(ALPHA,TWOB,EPS,ETAI,ACC)	SHELL066
	VB = -(2.0/(BETA**2))*FJ2(ALPHA,TWOB,ETAI,ACC)	SHELL067
	W = SU *T	SHELL068
	WA = SU*TA + T*SUA	SHELL069
	WB = SU*TB + T*SUB	SHELL070
C		SHELL071
C	TO FIND THE VALUES OF MU AND LAMDA	SHELL072
C		SHELL073
	CALL FMULAM ( FP, FQ, FM, FN, VMIN, WMAX, V, k, FMU, FLAM, NC )	SHELL074
C		SHELL075
	WRITE OUTPUT TAPE 6, 26, V, VA, VB, W, WA, WB, FMU, FLAM	SHELL076
	26 FORMAT (10X, 8F11.6 )	SHELL077
	IF (FMU) 50, 30, 30	SHELL078
	30 IF (FLAM) 50, 31, 31	SHELL079
C		SHELL080
C	TO CALCULATE FALPHA AND FBETA (THAT IS, ATIL AND BTIL)	SHELL081
C		SHELL082
	31 ATIL = FLAM * VA/ VMIN - FMU * WA/ WMAX	SHELL083
	BTIL = FLAM * VB/ VMIN - FMU * WB/ WMAX	SHELL084
	ATILS = ( ATIL*VA + BTIL*VB ) /VMIN	SHELL085
	BTILS = ( ATIL*WA + BTIL*WB ) / WMAX	SHELL086
	VO = V/VMIN	SHELL087
	WO = W/WMAX	SHELL088
C		SHELL089
C	PROGRAM TERMINATES WHEN QQ LESS THAN QEPS	SHELL090
C		SHELL091
	QQ = SQRT (ATIL**2 + BTIL**2)	SHELL092
	IF ( QQ-QEPS ) 48, 48, 35	SHELL093
C		SHELL094
C	TO DETERMINE STEP SIZE, CONTROL ON FKAP	SHELL095
C		SHELL096
	35 FKAPS = ETAC / SQRT (ATILS**2 + BTILS**2)	SHELL097
	IF (FKAPS - FKV) 36, 36, 37	SHELL098
	36 FKAP = FKAPS	SHELL099
	GO TO 40	SHELL100
	37 FKAP = FKV	SHELL101
	40 DA = FKAP * ATIL	SHELL102
	DB = FKAP * BTIL	SHELL103
	WRITE OUTPUT TAPE 6, 42, ALPHA, BETA, DA, DB, VO, WO, FKAP, ATIL,	SHELL104
	1 BTIL, ATILS, BTILS	SHELL105
	42 FORMAT (6F9.5, 5E12.5)	SHELL106
	ALPHA = ALPHA+ DA	SHELL107
	BETA = BETA + DB	SHELL108
	IF ( NCCNT - 50 ) 14, 50, 50	SHELL109
	48 WRITE OUTPUT TAPE 6, 49, ALPHA, BETA, VO, WO, ATIL, BTIL, ATILS,	SHELL110
	1 BTILS	SHELL111
	49 FORMAT ( 1H-, 8F13.5 /1H2 )	SHELL112
	50 CONTINUE	SHELL113
	69 CONTINUE	SHELL114
	CALL SYSTEM	SHELL115
	END	SHELL116

```

$      FORTRAN IBM
$      PUNCH OBJECT
SUBROUTINE FMULAM(P,Q,FM, FN, VMIN, WMAX, V, W, FMU, FLAM, NC)
1 VV = V/VMIN
2 WW = W/WMAX
3 IF (VV - 1.0) 4, 4, 10
4 IF (WW - 1.0) 10, 5, 5
5 WRITE OUTPUT TAPE 6, 6, VMIN, WMAX, VV, WW
6 FORMAT(7H VMIN = ,E12.6,8H WMAX = ,E12.6,6H V = ,E14.8,6H W = ,
1E14.8, // 5X, 43HTHE PRIMARY REQUIREMENTS CANNOT BE REACHED )
7 FMU = -1.0
8 FLAM = -1.0
9 RETURN
10 IF (V-1.0) 11, 5, 5
11 IF (VV - 1.0) 12, 12, 15
12 FMU = 0.0
13 FLAM = 1.0
14 IF (WW) 5, 5, 9
15 IF (WW) 5, 5, 16
16 IF (WW - 1.0) 20, 17, 17
17 FLAM = 0.0
18 FMU = 1.0
19 RETURN
20 FLAM = (( 1./VMIN - VV)**P)/ ((VV-1. )**Q)
21 FMU = (WW**FM)/((1.0 - WW)**FN)
22 GO TO (23,26), NC
23 TEMP = SQRT(FLAM**2 + FMU**2)
24 FLAM = FLAM/TEMP
25 FMU = FMU/TEMP
26 RETURN
END
SUBFML01
SUBFML02
SUBFML03
SUBFML04
SUBFML05
SUBFML06
SUBFML07
SUBFML08
SUBFML09
SUBFML10
SUBFML11
SUBFML12
SUBFML13
SUBFML14
SUBFML15
SUBFML16
SUBFML17
SUBFML18
SUBFML19
SUBFML20
SUBFML21
SUBFML22
SUBFML23
SUBFML24
SUBFML25
SUBFML26
SUBFML27
SUBFML28
SUBFML29

```

```

$      FORTRAN IBM
$PUNCH OBJECT
FUNCTION FJ1(P,Q,ACC)
1 ODD = 0.0
2 INT = 1
3 V = 1.0
4 EVEN = 0.0
  AREA1 = 0.0
  IF (Q) 19, 5, 6
5 ENDS = 2.0
  GO TO 7
6 ENDS = 1.0
7 H = 1.0/V
8 ODD = EVEN + ODD
9 X = H/2.
10 EVEN = 0.0
11 DO 13 I = 1, INT
12 EVEN = EVEN + ((1.0 - X**P)**Q)
13 X = X + H
14 AREA = (ENDS + 4.0*EVEN + 2.0*ODD)*H/6.0
15 R = ABSF(AREA1/AREA - 1.0) - ACC
16 IF (R) 25, 25, 17
17 IF (INT - 16384) 21, 19, 19
18 FORMAT (23H J1(P,Q) NOT CONVERGENT)
19 WRITE OUTPUT TAPF 6, 18
20 CALL SYSERR
21 AREA1 = AREA
22 INT = 2*INT
23 V = 2.0*V
24 GO TO 7
25 FJ1 = AREA
26 RETURN
END
FJ1 01
FJ1 02
FJ1 03
FJ1 04
FJ1 05
FJ1 06
FJ1 07
FJ1 08
FJ1 09
FJ1 10
FJ1 11
FJ1 12
FJ1 13
FJ1 14
FJ1 15
FJ1 16
FJ1 17
FJ1 18
FJ1 19
FJ1 20
FJ1 21
FJ1 22
FJ1 23
FJ1 24
FJ1 25
FJ1 26
FJ1 27
FJ1 28
FJ1 29
FJ1 30
FJ1 31

```

\$	FORTRAN IBM	
\$PUNCH	OBJECT	
	FUNCTION FJ2(P,Q,ETA1, ACC)	FJ2 01
1	DP = P	FJ2 02
	E= 1.-(1.0-ETA1)**P	FJ2 03
	OME = 1.0 -ETA1	FJ2 04
3	DEL = 1.0 - 1.0/DP	FJ2 05
5	F0 = 1.0	FJ2 06
6	F1 = (1.0 - 0.5*E)**(-DEL)	FJ2 07
7	F2 = (1.0 - E)**(-DEL)	FJ2 08
8	DQ = Q	FJ2 09
9	A = 4.0*F1 - F2 - 3.0*F0	FJ2 10
10	B = 2.0*F2 - 4.0*F1 + 2.0*F0	FJ2 11
11	T1 = DQ + 1.0	FJ2 12
12	T2 = DQ + 2.0	FJ2 13
13	T3 = DQ + 3.0	FJ2 14
14	EN = (ELOG (E)*(F0 + T1*(A/T2 + B/T3)) - (F0 + T1*T1*(A/(T2*T2)	FJ2 15
	1 + B/(T3*T3)))/T1)*(E**T1)/(T1*DP)	FJ2 16
15	ODD = 0.0	FJ2 17
16	INT = 1	FJ2 18
17	V = 1.0	FJ2 19
18	EVEN = 0.0	FJ2 20
	AREA1 = 0.0	FJ2 21
19	ENDS = ((1.0 - OME**DP)**DQ)*ELOG (1.0 - OME**DP)	FJ2 22
20	H = OME/V	FJ2 23
21	ODD = EVEN + ODD	FJ2 24
22	X = H/2.0	FJ2 25
23	EVEN = 0.0	FJ2 26
24	DO 26 I = 1, INT	FJ2 27
25	EVEN = EVEN + ((1.0 - X**DP)**DQ)*ELOG (1.0 - X**DP)	FJ2 28
26	X = X + H	FJ2 29
27	AREA = (ENDS + 4.0*EVEN + 2.0*ODD)*H/6.0	FJ2 30
28	R = ABSF(AREA1/AREA - 1.0) - ACC	FJ2 31
29	IF (R) 38, 38, 30	FJ2 32
30	IF (INT - 16384) 34, 31, 31	FJ2 33
31	WRITE OUTPUT TAPE 6, 32	FJ2 34
32	FORMAT (23H J2(P,Q) NOT CONVERGENT)	FJ2 35
33	CALL SYSERR	FJ2 36
34	AREA1 = AREA	FJ2 37
35	INT = 2*INT	FJ2 38
36	V = 2.0*V	FJ2 39
37	GO TO 20	FJ2 40
38	FJ2 = AREA + EN	FJ2 41
39	RETURN	FJ2 42
	END	FJ2 43

```

$      FORTRAN IBM
$PUNCH OBJECT
      FUNCTION FJ3(P,Q,EPS, ETAI, ACC)
1  DP = P
2  DQ = C - 1.0
3  T1 = DP + 1.0
4  T2 = T1/(DP + 2.0)
5  T3 = T1/(DP + 3.0)
6  FO = 1.0
7  E = EPS
8  F1 = (1.0 - (0.5*E)**DP)**DQ
9  F2 = (1.0 - E**DP)**DQ
10 A = 4.0*F1 - F2 - 3.0*FO
11 B = 2.0*F2 - 4.0*F1 + 2.0*FO
120EN = (ELOG (E)*(FO + T2*A + T3*B) - (FO + T2*T2*A + T3*T3*B)/T1)*(
1E**T1)/T1
13 DEL = 1.0/DP
   OME = 1.0-ETAI
   E = 1.0- OME**DP
15 T1 = DQ + 1.0
16 T2 = T1/(DQ + 2.0)
17 T3 = T1/(DQ + 3.0)
18 FO = 0.0
19 F1 = ((1.0 - 0.5*E)**DEL)*ELOG (1.0 - 0.5*E)
20 F2 = ((1.0 - E)**DEL)*ELOG (1.0 - E)
21 A = 4.0*F1 - F2 - 3.0*FO
22 B = 2.0*F2 - 4.0*F1 + 2.0*FO
23 EN = EN + (FO + T2*A + T3*B)*(E**T1)/(T1*DP*DP)
   E = EPS
240ENDS = ((1.0 - E**DP)**DQ)*(E**DP)*ELOG (E)
   1 + ((1.0 - CME**DP)**DQ)*(OME**DP)*FLCG (OME)
25 ODD = 0.0
26 INT = 1
27 V = 1.0
28 EVEN = 0.0
29 AREA1 = 0.0
30 H = (OME - E)/V
31 ODD = EVEN + CDD
32 X = E + H/2.
33 EVEN = 0.0
34 DO 36 I = 1, INT
35 EVEN = EVEN + ((1.0 - X**DP)**DQ)*(X**DP)*ELOG (X)
36 X = X + H
37 AREA = (ENDS + 4.0*EVEN + 2.0*ODD)*H/6.0
38 R = ABSF(AREA1/AREA - 1.0) - ACC
39 IF (R) 48, 48, 40
40 IF (INT - 16384) 44, 42, 42
41 FORMAT (23H J3(P,Q) NOT CONVERGENT)
42 WRITE OUTPUT TAPE 6, 41
43 CALL SYSCRR
44 AREA1 = AREA
45 INT = 2*INT
46 V = 2.0*V
47 GO TO 30
48 FJ3 = AREA + EN
49 RETURN
      END
      FJ3 01
      FJ3 02
      FJ3 03
      FJ3 04
      FJ3 05
      FJ3 06
      FJ3 07
      FJ3 08
      FJ3 09
      FJ3 10
      FJ3 11
      FJ3 12
      FJ3 13
      FJ3 14
      FJ3 15
      FJ3 16
      FJ3 17
      FJ3 18
      FJ3 19
      FJ3 20
      FJ3 21
      FJ3 22
      FJ3 23
      FJ3 24
      FJ3 25
      FJ3 26
      FJ3 27
      FJ3 28
      FJ3 29
      FJ3 30
      FJ3 31
      FJ3 32
      FJ3 33
      FJ3 34
      FJ3 35
      FJ3 36
      FJ3 37
      FJ3 38
      FJ3 39
      FJ3 40
      FJ3 41
      FJ3 42
      FJ3 43
      FJ3 44
      FJ3 45
      FJ3 46
      FJ3 47
      FJ3 48
      FJ3 49
      FJ3 50
      FJ3 51
      FJ3 52
      FJ3 53
      FJ3 54
      FJ3 55

```

```

$      FORTRAN IBM
$PUNCH OBJECT
      FUNCTION FJ4(P,Q,S,FK2, ETA1, ACC)
1  DG = S + Q - 1.0
2  T1 = DG + 1.0
3  T2 = T1/(DG + 2.0)
4  T3 = T1/(DG + 3.0)
      OME = 1.0 - ETA1
      E = 1.0 - OME**P
6  DP = P
8  DEL = -2.0*(1.0 - 1.0/DP)
      DQ = 2.0*(1.0 - Q)
      IF (Q - 1.0) 11, 10, 36
10  FO = SQRT (1.0/FK2 + 1.0)
      GO TO 12
11  FO = 1.0
12  DF = FK2
13  F1 = SQRT (((1.0 - 0.5*E)**DEL)*((0.5*E)**DQ)/FK2 + 1.0)
14  F2 = SQRT (((1.0 - E)**DEL)*(E**DQ)/FK2 + 1.0)
15  A = 4.0*F1 - F2 - 3.0*FO
16  B = 2.0*F2 - 4.0*F1 + 2.0*FO
17  EN = (FO + T2*A + T3*B)*(E**T1)*SQRT (DF)/(T1*DP)
18  ODD = 0.0
19  INT = 1
20  V = 1.0
21  EVEN = 0.0
      AREA1 = 0.0
      EE = 1.0
      IF (P - 1.0) 36, 22, 23
22  EE = SQRT (1.0 + FK2)
23  ENDS = EE + ((1.0 - OME**DP)**DG)*SQRT ((1.0 - OME**DP)**DQ + DF*
      1(OME**(2.0*DP - 2.0)))
24  DDP = 2.0*(DP - 1.0)
25  H = OME/V
26  ODD = EVEN + ODD
27  X = H/2.0
28  EVEN = 0.0
29  DO 31 I = 1, INT
30  EVEN = EVEN + ((1.0 - X**DP)**DG)*SQRT ((1.0 - X**DP)**DQ + DF*(X*
      1*DDP))
31  X = X + H
32  AREA = (ENDS + 4.0*EVEN + 2.0*ODD)*H/6.0
33  R = ABSF(AREA1/AREA - 1.0) - ACC
34  IF (R) 43, 43, 35
35  IF (INT - 16384) 39, 36, 36
36  WRITE OUTPUT TAPE 6, 37
37  FORMAT (25H J4(P,Q,S) NOT CONVERGENT)
38  CALL SYSERR
39  AREA1 = AREA
40  INT = 2*INT
41  V = 2.0*V
42  GO TO 25
43  FJ4 = AREA + EN
44  RETURN
      END
      FJ4 01
      FJ4 02
      FJ4 03
      FJ4 04
      FJ4 05
      FJ4 06
      FJ4 07
      FJ4 08
      FJ4 09
      FJ4 10
      FJ4 11
      FJ4 12
      FJ4 13
      FJ4 14
      FJ4 15
      FJ4 16
      FJ4 17
      FJ4 18
      FJ4 19
      FJ4 20
      FJ4 21
      FJ4 22
      FJ4 23
      FJ4 24
      FJ4 25
      FJ4 26
      FJ4 27
      FJ4 28
      FJ4 29
      FJ4 30
      FJ4 31
      FJ4 32
      FJ4 33
      FJ4 34
      FJ4 35
      FJ4 36
      FJ4 37
      FJ4 38
      FJ4 39
      FJ4 40
      FJ4 41
      FJ4 42
      FJ4 43
      FJ4 44
      FJ4 45
      FJ4 46
      FJ4 47
      FJ4 48
      FJ4 49
      FJ4 50
      FJ4 51
      FJ4 52
      FJ4 53

```

```

$      FORTRAN IBM
$PUNCH OBJECT
      FUNCTION FJ5(P,Q,S,FK2,CTAI,ACC)
      OME = 1.0 - ETAI
      2 G = S - Q - 1.0
      3 QQ = 2.0*(1.0 - Q)
      4 DEL = 1.0 - 1.0/P
      TDEL = 2.0*DEL
      FK = FK2
      IF (Q - 1.0) 6, 5, 35
      5 FK = 1.0 + FK
      6 F0 = 1.0/SQRT (FK)
      EPS = 1.0 - OME**P
      7 TPS = 0.5*EPS
      8 F1 = ((1.0 - TPS)**DEL)/SQRT (TPS**QQ + FK2*((1.0 - TPS)**TDEL))
      9 F2 = ((1.0 - EPS)**DEL)/SQRT (EPS**QQ + FK2*((1.0 - EPS)**TDEL))
      10 A = 4.0*F1 - F2 - 3.0*F0
      11 B = 2.0*F2 - 4.0*F1 + 2.0*F0
      12 T1 = G + 1.0
      13 T2 = T1/(G + 2.0)
      14 T3 = T1/(G + 3.0)
      15 EN = (F0 + T2*A + T3*B)*(EPS**T1)/(T1*P)
      16 ODD = 0.0
      17 INT = 1
      18 V = 1.0
      19 EVEN = 0.0
      AREA1 = 0.0
      EE = 0.0
      IF (P - 1.0) 35, 20, 21
      20 EE = 1.0/SQRT (1.0 + FK2)
      21 ENDS = SQRT ((1.0 - OME**P)**QQ + FK2*(OME**(2.0*P - 2.0)))
      22 PP = 2.0*(P - 1.0)
      23 ENDS = (OME**PP)*((1.0 - OME**P)**G)/ENDS + EE
      24 H = OME/V
      25 ODD = EVEN + ODD
      26 X = H/2.0
      27 EVEN = 0.0
      28 DO 30 I = 1, INT
      29O EVEN = EVEN + (X**PP)*((1.0 - X**P)**G)/SQRT ((1.0 - X**P)**QQ +
      1 FK2*(X**PP))
      30 X = X + H
      31 AREA = (ENDS + 4.0*EVEN + 2.0*ODD)*H/6.0
      32 R = ABSF(AREA1/AREA - 1.0) - ACC
      33 IF (R) 42, 42, 34
      34 IF (INT - 16384) 38, 35, 35
      35 WRITE OUTPUT TAPE 6, 36
      36 FORMAT (25H J5(P,Q,S) NCT CONVERGENT)
      37 CALL SYSERR
      38 AREA1 = AREA
      39 INT = 2*INT
      40 V = 2.0*V
      41 GO TO 24
      42 FJ5 = AREA + EN
      43 RETURN
      END
      FJ5 01
      FJ5 02
      FJ5 03
      FJ5 04
      FJ5 05
      FJ5 06
      FJ5 07
      FJ5 08
      FJ5 09
      FJ5 10
      FJ5 11
      FJ5 12
      FJ5 13
      FJ5 14
      FJ5 15
      FJ5 16
      FJ5 17
      FJ5 18
      FJ5 19
      FJ5 20
      FJ5 21
      FJ5 22
      FJ5 23
      FJ5 24
      FJ5 25
      FJ5 26
      FJ5 27
      FJ5 28
      FJ5 29
      FJ5 30
      FJ5 31
      FJ5 32
      FJ5 33
      FJ5 34
      FJ5 35
      FJ5 36
      FJ5 37
      FJ5 38
      FJ5 39
      FJ5 40
      FJ5 41
      FJ5 42
      FJ5 43
      FJ5 44
      FJ5 45
      FJ5 46
      FJ5 47
      FJ5 48
      FJ5 49
      FJ5 50
      FJ5 51
      FJ5 52
      FJ5 53

```

\$	FORTRAN IBM	
\$PUNCH	OBJECT	
	FUNCTION FJ6(P,Q,R,S,FK2, ETAI, ACC)	FJ6 01
1	TQ = 2.0*(1.0 - Q)	FJ6 02
2	TP = 2.0*(P - 1.0)	FJ6 03
3	TR = 2.0*(R - 1.0)	FJ6 04
4	G = S - Q - 1.0	FJ6 05
5	TD = 2.0*(1.0 - 1.0/P)	FJ6 06
	EPS = 1.0 - (1.0 - ETAI)**P	FJ6 07
6	TPS = 0.5*EPS	FJ6 08
7	TT = 2.0*R/P - 1.0 - 1.0/P	FJ6 09
8	T1 = G + 1.0	FJ6 10
9	T2 = T1/(G + 2.0)	FJ6 11
	T3 = T1/(G + 3.0)	FJ6 12
	FK = FK2	FJ6 13
	IF (Q - 1.0) 11, 10, 39	FJ6 14
10	FK = 1.0 + FK	FJ6 15
11	FO = 1.0/SQRT (FK)	FJ6 16
12	OMW = 1.0 - TPS	FJ6 17
13	F1 = (OMW**TT)/SQRT (TPS**TQ + FK2*(OMW**TD))	FJ6 18
14	OMW = 1.0 - EPS	FJ6 19
15	F2 = (OMW**TT)/SQRT (EPS**TQ + FK2*(OMW**TD))	FJ6 20
	OME = 1.0 - ETAI	FJ6 21
17	A = 4.0*F1 - F2 - 3.0*FO	FJ6 22
18	B = 2.0*F2 - 4.0*F1 + 2.0*FO	FJ6 23
19	ODEN = (ELOG (EPS)*(FO + T2*A + T3*B) - (FO + T2*T2*A + T3*T3*R)/T1)	FJ6 24
	1*(EPS**T1)/(T1*P)	FJ6 25
20	ODD = 0.0	FJ6 26
21	INT = 1	FJ6 27
22	V = 1.0	FJ6 28
23	EVEN = 0.0	FJ6 29
24	AREAL = 0.0	FJ6 30
25	OMW = 1.0 - OME**P	FJ6 31
26	ENDS = ((OME**TR)*(OMW**G)/SQRT (OMW**TQ + FK2*(OME**TP)))*ELOG (OMW	FJ6 32
	1W)	FJ6 33
27	H = OME/V	FJ6 34
28	CDD = EVEN + CDD	FJ6 35
29	X = H/2.0	FJ6 36
30	EVEN = 0.0	FJ6 37
31	DO 34 I = 1, INT	FJ6 38
32	OMW = 1.0 - X**P	FJ6 39
33	EVEN = EVEN + (((X**TR)*(OMW**G)/SQRT (OMW**TQ + FK2*(X**TP)))*ELOG	FJ6 40
	1(OMW)	FJ6 41
34	X = X + H	FJ6 42
35	AREA = (ENDS + 4.0*EVEN + 2.0*ODD)*H/6.0	FJ6 43
36	RR = ABSF(AREAL/AREA - 1.0) - ACC	FJ6 44
37	IF (RR) 46, 46, 38	FJ6 45
38	IF (INT - 16384) 42, 39, 39	FJ6 46
39	WRITE OUTPUT TAPE 6, 40	FJ6 47
40	FORMAT (27H J6(P,Q,R,S) NOT CONVERGENT)	FJ6 48
41	CALL SYSERR	FJ6 49
42	AREAL = AREA	FJ6 50
43	INT = 2*INT	FJ6 51
44	V = 2.0*V	FJ6 52
45	GO TO 27	FJ6 53
46	FJ6 = AREA + EN	FJ6 54
47	RETURN	FJ6 55
	END	FJ6 56

```

$      FORTRAN IBM
$PUNCH OBJECT
      FUNCTION FJ7(P,Q,S,T,FK2,EPS, ETAI, ACC)
1  TQ = 2.0*(1.0 - Q)
2  TP = 2.0*(P - 1.0)
3  G = S - Q - 2.0
4  T1 = T + 1.0
5  T2 = T1/(T + 2.0)
   T3 = T1/(T + 3.0)
   IF (P - 1.0) 53, 6, 7
6  F0 = 1.0/SQRT (1.0 + FK2)
   GO TO 8
7  F0 = 1.0
8  TPS = 0.5*EPS
9  OMW = 1.0 - TPS**P
10 F1 = (OMW**G)/SQRT (OMW**TQ + FK2*(TPS**TP))
11 OMW = 1.0 - EPS**P
12 F2 = (OMW**G)/SQRT (OMW**TQ + FK2*(EPS**TP))
13 A = 4.0*F1 - F2 - 3.0*F0
14 B = 2.0*F2 - 4.0*F1 + 2.0*F0
15 TT = F0 + T2*A + T3*B
16 TU = F0 + T2*T2*A + T3*T3*B
17 EN = (ELOG (EPS)*TT - TU/T1)*(EPS**T1)/T1
   OME = 1.0 - ETAI
19 T4 = 2.0*(1.0 - 1.0/P)
20 T5 = T/P - 1.0 + 1.0/P
21 T1 = G + 2.0
22 T2 = T1/(G + 3.0)
   T3 = T1/(G + 4.0)
   FK = FK2
   IF (Q - 1.0) 24, 23, 53
23 FK = 1.0 + FK
24 F0 = -1.0/SQRT (FK)
   E = 1.0 - OME**P
   TPS = 0.5 * E
   OMW = 1. - TPS
27 F1 = (OMW**T5)/SQRT (TPS**TQ + FK2*(OMW**T4)) * ELOG(OMW) /TPS
28 OMW = 1.0 - E
   F2 = (OMW**T5)/SQRT ( E **TQ + FK2*(OMW**T4)) * ELOG(OMW) /E
30 A = 4.0*F1 - F2 - 3.0*F0
31 B = 2.0*F2 - 4.0*F1 + 2.0*F0
32 EN = EN + (F0 + T2*A + T3*B)*( E **T1)/(T1*P*P)
33 ODD = 0.0
34 INT = 1
35 V = 1.0
36 EVEN = 0.0
37 AREA1 = 0.0
   OMW = 1.0 - EPS**P
38 ENDS = (EPS**T)*(OMW**G)*ELOG (EPS)/SQRT (OMW**TQ + FK2*(EPS**TP))
39 O = 1.0 - OME**P
40 ENDS=ENDS+(OME**T)*(O**G)*ELOG (OME)/SQRT (O**TQ+FK2*(OME**TP))
41 H = (OME - EPS)/V
42 ODD = EVEN + ODD
43 X = EPS + H/2.0
44 EVEN = 0.0
45 DO 48 I = 1, INT
46 O = 1.0 - X**P
47 EVEN = EVEN + (X**T)*(O**G)*ELOG (X)/SQRT (O**TQ + FK2*(X**TP))
48 X = X + H
49 AREA = (ENDS + 4.0*EVEN + 2.0*ODD)*H/6.0
50 R = ABSF(AREA1/AREA - 1.0) - ACC
51 IF (R) 60, 60, 52
52 IF (INT - 16384) 56, 53, 53
53 WRITE OUTPUT TAPE 6, 54
54 FORMAT (27H J7(P,Q,S,T) NOT CONVERGENT)
55 CALL SYSERR
56 AREA1 = AREA
57 INT = 2*INT
58 V = 2.0*V
59 GO TO 41
60 FJ7 = AREA + EN
61 RETURN
   END
      FJ7 01
      FJ7 02
      FJ7 03
      FJ7 04
      FJ7 05
      FJ7 06
      FJ7 07
      FJ7 08
      FJ7 09
      FJ7 10
      FJ7 11
      FJ7 12
      FJ7 13
      FJ7 14
      FJ7 15
      FJ7 16
      FJ7 17
      FJ7 18
      FJ7 19
      FJ7 20
      FJ7 21
      FJ7 22
      FJ7 23
      FJ7 24
      FJ7 25
      FJ7 26
      FJ7 27
      FJ7 28
      FJ7 29
      FJ7 30
      FJ7 31
      FJ7 32
      FJ7 33
      FJ7 34
      FJ7 35
      FJ7 36
      FJ7 37
      FJ7 38
      FJ7 39
      FJ7 40
      FJ7 41
      FJ7 42
      FJ7 43
      FJ7 44
      FJ7 45
      FJ7 46
      FJ7 47
      FJ7 48
      FJ7 49
      FJ7 50
      FJ7 51
      FJ7 52
      FJ7 53
      FJ7 54
      FJ7 55
      FJ7 56
      FJ7 57
      FJ7 58
      FJ7 59
      FJ7 60
      FJ7 61
      FJ7 62
      FJ7 63
      FJ7 64
      FJ7 65
      FJ7 66
      FJ7 67
      FJ7 68
      FJ7 69
      FJ7 70
      FJ7 71

```



## REFERENCES

1. "Geometrical and Inertial Properties of a Class of Thin Shells of Revolution", by  
Will J. Worley and Han-chung Wang  
National Aeronautics and Space Administration, Grant No. NsG-434,  
N.A.S.A. Contractor Report CR-89, September, 1964, 208 pages.
2. "Geometrical and Inertial Properties of a Class of Thin Shells of a General Type" by  
Will J. Worley and Han-chung Wang  
National Aeronautics and Space Administration, Grant No. NsG-434,  
Supplement No. 1, NASA Contractor Report CR-271, Aug. 1965, 67 pages.
3. "Minimum Weight Design of Cylindrical Shells" by  
Walter Freiberger  
Journal of Applied Mechanics, Vol. 23, No. 4., Trans. ASME,  
Dec. 1956, pp. 576-580.
4. "On the Optimum Design of Shells" by  
R. T. Shield  
Journal of Applied Mechanics, Vol. 27, No. 2, Trans. ASME,  
June 1960, pp. 316-322.
5. "Numerical Analysis of Unsymmetrical Bending of Shells of Revolution" by  
B. Budiansky and P. P. Radkowski  
AIAA Journal, Vol. 1, No. 8, Aug. 1963, pp. 1833-1842.
6. "Numerical Analysis of Equations of Thin Shell of Revolution" by  
P. P. Radkowski, R. M. Davis and M. R. Bolduc,  
American Rocket Society Journal, Vol. 32, No. 1, Jan. 1962, pp.36-41.
7. "Analysis of Shells of Revolution Subjected to Symmetrical and Nonsymmetrical  
Loads" by  
A. Kalnins  
Journal of Applied Mechanics, Vol. 31, No. 3, Trans. ASME  
Sept. 1964, pp. 467-476.

TABLE I ITERATION RESULTS FOR  $\lambda=\lambda_1, \mu=\mu_1, b/a=1, \alpha_0=1.5, \beta_0=1.5$

Step	$\alpha$	$\beta$	$\frac{V}{2\pi ab^2}$	$\frac{W(\frac{\sigma_0}{p_0})}{4\pi ab^2 \rho g}$	$\frac{t}{b}(\frac{\sigma_0}{p_0})$	$F_\alpha$	$F_\beta$	$\Delta\alpha$	$\Delta\beta$
1	1.5000	1.5000	.53743	.43810	.5000	.29499	.24269	.0885	.0728
2	1.5885	1.5728	.56271	.44979	.5000	.27202	.22630	.0816	.0679
3	1.6701	1.6407	.58448	.46001	.5000	.25255	.21206	.0758	.0636
4	1.7459	1.7043	.60343	.46903	.5000	.26893	.22758	.2228	.1885
5	1.9686	1.8928	.65276	.49306	.5000	.83353	.72064	.2501	.2162
6	2.2187	2.1090	.69862	.52328	.5069	-.28840	-.28654	-.0865	-.0860
7	2.1322	2.0231	.68274	.51165	.5035	-.05475	-.08112	-.0164	-.0243
8	2.1189	1.9915	.67861	.50850	.5025	.01648	-.01889	.0049	-.0057
9	2.1288	1.9802	.67860	.50846	.5024	.01672	-.01885	.0050	-.0057
10	2.1389	1.9689	.67859	.50841	.5024	.01689	-.01889	.0051	-.0057
11	2.1490	1.9575	.67859	.50837	.5023	.01705	-.01896	.0051	-.0057
12	2.1593	1.9461	.67858	.50832	.5023	.01721	-.01899	.0052	-.0057
13	2.1696	1.9347	.67856	.50827	.5022	.01756	-.01886	.0053	-.0057
14	2.1801	1.9233	.67856	.50823	.5022	.01758	-.01905	.0053	-.0057
15	2.1998	1.9107	.67935	.50875	.5023	.00421	-.03187	.0021	-.0159
16	2.2151	1.8891	.67884	.50830	.5021	.01305	-.02400	.0065	-.0120
17	2.2323	1.8690	.67865	.50807	.5020	.01655	-.02106	.0083	-.0105
18	2.2505	1.8494	.67857	.50793	.5019	.01806	-.01998	.0090	-.0100
19	2.2692	1.8299	.67853	.50783	.5017	.01883	-.01960	.0094	-.0098
20	2.2883	1.8105	.67851	.50774	.5016	.01927	-.01957	.0096	-.0098
21	2.3078	1.7911	.67849	.50764	.5015	.01971	-.01951	.0099	-.0098
22	2.3277	1.7716	.67847	.50754	.5014	.02008	-.01952	.0100	-.0098
23	2.3478	1.7521	.67846	.50745	.5013	.02044	-.01955	.0102	-.0098
24	2.3684	1.7326	.67844	.50735	.5012	.02080	-.01959	.0104	-.0098
25	2.3893	1.7130	.67843	.50725	.5011	.02116	-.01963	.0106	-.0098
26	2.4105	1.6933	.67842	.50715	.5010	.02153	-.01966	.0108	-.0098
27	2.4321	1.6737	.67840	.50704	.5008	.02189	-.01968	.0110	-.0098
28	2.4541	1.6540	.67838	.50693	.5007	.02227	-.01970	.0111	-.0099
29	2.4765	1.6343	.67837	.50681	.5005	.02276	-.01955	.0114	-.0098
30	2.4992	1.6146	.67836	.50670	.5004	.02310	-.01967	.0116	-.0098
31	2.5224	1.5948	.67834	.50659	.5003	.02345	-.01974	.0117	-.0099

TABLE II ITERATION RESULTS FOR  $\lambda=\lambda_1, \mu=\mu_1, b/a=1, \alpha_0=2.0, \beta_0=3.0$

Step	$\alpha$	$\beta$	$\frac{V}{2\pi ab^2}$	$\frac{W(\frac{\sigma_0}{p_0})}{4\pi ab^2 \rho g}$	$\frac{t}{b}(\frac{\sigma_0}{p_0})$	$F_\alpha$	$F_\beta$	$\Delta\alpha$	$\Delta\beta$
1	2.0000	3.0000	.73915	.55381	.5151	-.99584	-.64496	-.2988	-.1935
2	1.7013	2.8065	.69653	.52338	.5075	-.32702	-.21005	-.0981	-.0630
3	1.6031	2.7435	.68018	.51171	.5042	-.02135	-.03251	-.0064	-.0098
4	1.5967	2.7337	.67866	.51062	.5040	.01177	-.01391	.0035	-.0042
5	1.6003	2.7296	.67881	.51071	.5039	.00845	-.01582	.0025	-.0048
6	1.6028	2.7248	.67879	.51069	.5039	.00887	-.01562	.0027	-.0047
7	1.6055	2.7201	.67879	.51068	.5039	.00886	-.01566	.0027	-.0047
8	1.6108	2.7107	.67879	.51066	.5039	.00893	-.01570	.0027	-.0047
9	1.6162	2.7013	.67879	.51064	.5039	.00900	-.01574	.0027	-.0047
10	1.6270	2.6824	.67879	.51060	.5039	.00914	-.01582	.0027	-.0047
11	1.6381	2.6634	.67876	.51055	.5039	.00955	-.01574	.0029	-.0047
12	1.6493	2.6443	.67877	.51051	.5039	.00943	-.01598	.0028	-.0048
13	1.6607	2.6251	.67877	.51047	.5039	.00958	-.01606	.0029	-.0048
14	1.6723	2.6058	.67876	.51042	.5039	.00973	-.01615	.0029	-.0048
15	1.6840	2.5864	.67876	.51038	.5038	.00989	-.01623	.0030	-.0049
16	1.6960	2.5668	.67876	.51033	.5038	.01005	-.01632	.0030	-.0049
17	1.7081	2.5472	.67875	.51028	.5038	.01022	-.01640	.0031	-.0049
18	1.7204	2.5275	.67875	.51023	.5038	.01039	-.01649	.0031	-.0050
19	1.7330	2.5077	.67875	.51018	.5038	.01056	-.01658	.0032	-.0050
20	1.7457	2.4877	.67874	.51012	.5037	.01074	-.01667	.0032	-.0050
21	1.7587	2.4677	.67874	.51007	.5037	.01093	-.01676	.0033	-.0050
22	1.7719	2.4475	.67874	.51001	.5037	.01112	-.01685	.0033	-.0051
23	1.7853	2.4273	.67873	.50996	.5036	.01131	-.01694	.0034	-.0051
24	1.7989	2.4071	.67873	.50990	.5036	.01147	-.01704	.0034	-.0051
25	1.8128	2.3866	.67872	.50984	.5036	.01171	-.01711	.0035	-.0051
26	1.8269	2.3660	.67872	.50978	.5035	.01192	-.01721	.0036	-.0052
27	1.8412	2.3453	.67870	.50972	.5035	.01205	-.01717	.0036	-.0052
28	1.8558	2.3247	.67870	.50966	.5035	.01251	-.01728	.0038	-.0052
29	1.8708	2.3038	.67870	.50958	.5034	.01277	-.01734	.0038	-.0052
30	1.8860	2.2828	.67871	.50952	.5034	.01280	-.01758	.0038	-.0053

TABLE III ITERATION RESULTS FOR  $\lambda=\lambda_1, \mu=\mu_1, b/a=2, \alpha_0=1.5, \beta_0=1.5$

Step	$\alpha$	$\beta$	$\frac{V}{2\pi ab^2}$	$\frac{W(\frac{\sigma_0}{p_0})}{4\pi ab^2 \rho g}$	$\frac{t}{b} (\frac{\sigma_0}{p_0})$	$F_\alpha$	$F_\beta$	$\Delta\alpha$	$\Delta\beta$
1	1.5000	1.5000	.53743	.62673	.5000	.29499	.24269	.0885	.0728
2	1.5885	1.5728	.56271	.63813	.5000	.27202	.22630	.0816	.0679
3	1.6701	1.6407	.58448	.64838	.5000	.25255	.21206	.0758	.0636
4	1.7459	1.7043	.60343	.65763	.5000	26.8590	22.7980	.2312	.1963
5	1.9771	1.9006	.65449	.68397	.5000	.72402	.70662	.2172	.2120
6	2.1944	2.1126	.69661	.70714	.5000	-.06577	.03817	-.0197	.0115
7	2.1303	2.1692	.69507	.70564	.5000	-.04389	.05289	-.0132	.0159
8	2.0792	2.2325	.69515	.70509	.5000	-.04058	.05212	-.0122	.0156
9	2.0317	2.2945	.69528	.70462	.5000	-.03797	.05084	-.0114	.0153
10	1.9872	2.3549	.69540	.70419	.5000	-.03561	.04950	-.0107	.0149
11	1.9455	2.4137	.69553	.70380	.5000	-.03342	.04814	-.0100	.0144
12	1.9063	2.4709	.69566	.70343	.5000	-.03135	.04680	-.0094	.0140
13	1.8696	2.5264	.69579	.70310	.5000	-.02940	.04543	-.0088	.0136
14	1.8012	2.6200	.69601	.70259	.5000	-.02680	.04309	-.0161	.0259
15	1.7494	2.7211	.69626	.70210	.5000	-.02346	.04044	-.0141	.0243
16	1.6956	2.8159	.69650	.70169	.5000	-.02089	.03796	-.0125	.0228
17	1.6475	2.9049	.69673	.70135	.5000	-.01870	.03561	-.0112	.0214
18	1.6043	2.9883	.69695	.70107	.5000	-.01680	.03339	-.0101	.0200
19	1.5656	3.0666	.69716	.70084	.5000	-.01515	.03132	-.0091	.0188
20	1.5305	3.1400	.69736	.70065	.5000	-.01370	.02938	-.0082	.0176
21	1.5065	3.1920	.69751	.70053	.5000	-.01273	.02801	-.0076	.0168
22	1.4770	3.2577	.69769	.70040	.5000	-.01159	.02628	-.0070	.0158
23	1.4502	3.3195	.69789	.70030	.5000	-.01091	.02453	-.0065	.0147
24	1.4256	3.3773	.69803	.70020	.5000	-.00966	.02317	-.0058	.0139
25	1.4032	3.4316	.69816	.70013	.5000	-.00884	.02178	-.0053	.0131
26	1.3826	3.4827	.69835	.70007	.5000	-.00817	.02043	-.0049	.0123
27	1.3636	3.5306	.69849	.70002	.5000	-.00747	.01923	-.0045	.0115
28	1.3463	3.5757	.69862	.69999	.5000	-.00688	.01809	-.0041	.0109
29	1.3302	3.6182	.69875	.69997	.5000	-.00641	.01702	-.0038	.0102
30	1.3154	3.6582	.69887	.69996	.5000	-.00590	.01601	-.0035	.0096
31	1.3017	3.6957	.69898	.69995	.5000	-.00544	.01509	-.0033	.0091

TABLE IV ITERATION RESULTS FOR  $\lambda=\lambda_1, \mu=\mu_1, b/a=2, \alpha_0=2.0, \beta_0=3.0$

Step	$\alpha$	$\beta$	$\frac{V}{2\pi ab^2}$	$\frac{W(\frac{\sigma_0}{p_0})}{4\pi ab^2 \rho g}$	$t \frac{\sigma_0}{b(p_0)}$	$F_\alpha$	$F_\beta$	$\Delta\alpha$	$\Delta\beta$
1	2.0000	3.0000	.73915	.72761	.5000	-.46258	-.21858	-.1388	-.0656
2	1.8612	2.9344	.72206	.71693	.5000	-.31011	-.12583	-.0930	-.0378
3	1.7682	2.8967	.70998	.70953	.5000	-.18587	-.05414	-.0558	-.0162
4	1.7124	2.8804	.70272	.70513	.5000	-.10019	-.00669	-.0301	-.0020
5	1.6824	2.8784	.69912	.70294	.5000	-.05330	.01847	-.0160	.0055
6	1.6664	2.8840	.69761	.70199	.5000	-.03224	.02937	-.0097	.0088
7	1.6567	2.8928	.69704	.70160	.5000	-.02379	.03344	-.0071	.0100
8	1.6496	2.9028	.69685	.70143	.5000	-.02050	.03476	-.0062	.0104
9	1.6434	2.9132	.69680	.70135	.5000	-.01915	.03506	-.0058	.0105
10	1.6377	2.9238	.69680	.70130	.5000	-.01851	.03498	-.0056	.0105
11	1.6161	2.9653	.69689	.70115	.5000	-.01735	.03398	-.0052	.0102
12	1.5957	3.0056	.69700	.70102	.5000	-.01646	.03292	-.0049	.0099
13	1.5763	3.0446	.69711	.70090	.5000	-.01563	.03188	-.0047	.0096
14	1.5579	3.0825	.69721	.70080	.5000	-.01485	.03088	-.0045	.0093
15	1.5361	3.1281	.69733	.70068	.5000	-.01392	.02969	-.0084	.0178
16	1.5039	3.1977	.69752	.70052	.5000	-.01263	.02786	-.0076	.0167
17	1.4746	3.2630	.69770	.70039	.5000	-.01148	.02615	-.0069	.0157
18	1.4480	3.3243	.69790	.70028	.5000	-.01048	.02455	-.0063	.0147
19	1.4237	3.3818	.69804	.70019	.5000	-.00958	.02306	-.0058	.0138
20	1.4014	3.4359	.69820	.70012	.5000	-.00878	.02167	-.0053	.0130
21	1.3809	3.4867	.69840	.70007	.5000	-.00811	.02033	-.0049	.0122

TABLE V ITERATION RESULTS FOR  $\lambda=\lambda_2, \mu=\mu_2, b/a=1$

Step	$\alpha$	$\beta$	$\frac{V}{2\pi ab^2}$	$\frac{W(\frac{\sigma_0}{p_0})}{4\pi ab^2 \rho g}$	$\frac{t(\frac{\sigma_0}{p_0})}{b}$	$F_\alpha$	$F_\beta$	$\Delta\alpha$	$\Delta\beta$
1	3.0000	3.0000	.80611	.60146	.5238	-.05923	-.05224	-.1777	-.1567
2	2.8223	2.8433	.78954	.58952	.5219	-.06529	-.05730	-.1959	-.1719
3	2.6265	2.6714	.76878	.57453	.5192	-.07086	-.06187	-.2126	-.1856
4	2.4139	2.4858	.74268	.55566	.5153	-.07227	-.06293	-.2168	-.1888
5	2.1971	2.2970	.71143	.53294	.5097	-.05642	-.04991	-.1693	-.1497
6	2.0278	2.1473	.68281	.51206	.5039	-.00851	-.01119	-.0255	-.0336
7	2.0023	2.1137	.67719	.50794	.5026	.00605	.00053	.0182	.0016
8	2.0205	2.1153	.67928	.50944	.5030	.00046	-.00402	.0014	-.0121
1	2.5000	2.5000	.75000	.56092	.5164	-.07211	-.06410	-.0721	-.0641
2	2.4279	2.4359	.74052	.55404	.5149	-.07125	-.06341	-.0713	-.0634
3	2.3566	2.3725	.73062	.54684	.5132	-.06854	-.06117	-.0685	-.0612
4	2.2881	2.3113	.72056	.53951	.5114	-.06330	-.05683	-.0633	-.0568
5	2.2248	2.2545	.71074	.53233	.5095	-.05507	-.05002	-.0551	-.0500
6	2.1697	2.2045	.70172	.52577	.5077	-.04418	-.04101	-.0442	-.0410
7	2.1256	2.1635	.69414	.52022	.5062	-.03206	-.03100	-.0321	-.0310
8	2.0935	2.1325	.68835	.51599	.5049	-.02082	-.02172	-.0208	-.0217
9	2.0727	2.1108	.68437	.51307	.5041	-.01202	-.01445	-.0120	-.0145
10	2.0607	2.0963	.68188	.51123	.5035	-.00605	-.00952	-.0061	-.0095
11	2.0546	2.0868	.68041	.51015	.5032	-.00239	-.00650	-.0024	-.0065
1	4.0000	2.0000	.80000	.59625	.5212	-.04804	-.07411	-.0480	-.0741
2	3.9520	1.9259	.79348	.59140	.5203	-.04939	-.07776	-.0494	-.0778
3	3.9026	1.8481	.78631	.58605	.5192	-.05070	-.08166	-.0507	-.0817
4	3.8519	1.7665	.77837	.58011	.5178	-.05189	-.08576	-.0519	-.0858
5	3.8000	1.6807	.76958	.57350	.5163	-.05282	-.08990	-.0528	-.0899
6	3.7472	1.5908	.75982	.56610	.5144	-.05324	-.09376	-.0532	-.0938
7	3.6939	1.4971	.74898	.55782	.5122	-.05273	-.09669	-.0527	-.0967
8	3.6412	1.4004	.73704	.54864	.5096	-.05063	-.09744	-.0506	-.0974
9	3.5906	1.3029	.72414	.53862	.5065	-.04591	-.09385	-.0459	-.0939
10	3.5447	1.2091	.71082	.52814	.5030	-.03744	-.08295	-.0374	-.0830
11	3.5072	1.1261	.69826	.51874	.4994	-.00550	-.00138	-.0055	-.0014

TABLE VI ITERATION RESULTS FOR  $\lambda=\lambda_3$ ,  $\mu=\mu_3$ ,  $b/a=1$

Step	$\alpha$	$\beta$	$\frac{V}{2\pi ab^2}$	$\frac{W(\frac{\sigma_0}{p_0})}{4\pi ab^2 \rho g}$	$\frac{t}{b}(\frac{\sigma_0}{p_0})$	$F_\alpha$	$F_\beta$	$\Delta\alpha$	$\Delta\beta$
1	1.5000	1.5000	.53743	.43810	.5000	.29499	.24269	.1838	.1512
2	1.6838	1.6512	.58787	.46162	.5000	.24949	.20988	.2121	.1784
3	1.8959	1.8296	.63745	.48553	.5000	.10529	.09039	.2106	.1808
4	2.1064	2.0104	.67902	.50889	.5026	.03332	.02410	.0667	.0482
5	2.1731	2.0586	.68997	.51692	.5050	.02556	.01793	.0511	.0359
6	2.2242	2.0944	.69793	.52274	.5067	.02035	.01376	.0407	.0275
7	2.3259	2.1606	.71246	.53335	.5100	.01173	.00680	.0235	.0136
8	2.3868	2.1937	.72019	.53898	.5110	.00761	.00342	.0152	.0069
9	2.4274	2.2100	.72473	.54228	.5118	.00532	.00153	.0107	.0031
10	2.4380	2.2131	.72584	.54306	.5120	.00479	.00109	.0096	.0022
1	2.5000	3.0000	.77759	.58111	.5206	-.01458	-.01210	-.0292	-.0242
2	2.4709	2.9758	.77447	.57887	.5202	-.01373	-.01149	-.0275	-.0230
3	2.4434	2.9528	.77145	.57670	.5198	-.01286	-.01088	-.0257	-.0218
4	2.4177	2.9311	.76856	.57463	.5194	-.01199	-.01027	-.0240	-.0205
5	2.3937	2.9105	.76581	.57265	.5190	-.01112	-.00966	-.0222	-.0193
6	2.3715	2.8912	.76320	.57077	.5187	-.01025	-.00906	-.0205	-.0181
7	2.3150	2.8403	.75626	.56578	.5177	-.00778	-.00736	-.0156	-.0147
8	2.2728	2.7992	.75070	.56178	.5168	-.00562	-.00590	-.0112	-.0118
9	2.2428	2.7663	.74642	.55869	.5162	-.00385	-.00471	-.0077	-.0094
10	2.2284	2.7482	.74419	.55708	.5158	-.00288	-.00406	-.0058	-.0081
1	3.0000	2.0000	.75000	.56001	.5151	-.00521	-.00949	-.0104	-.0190
2	2.9896	1.9810	.74796	.55847	.5148	-.00455	-.00885	-.0091	-.0177
3	2.9805	1.9633	.74606	.55707	.5144	-.00393	-.00825	-.0079	-.0165
4	2.9726	1.9468	.74430	.55576	.5141	-.00334	-.00767	-.0067	-.0153
5	2.9603	1.9172	.74123	.55346	.5135	-.00229	-.00663	-.0046	-.0133
6	2.9521	1.8916	.73869	.55154	.5130	-.00140	-.00575	-.0028	-.0115
7	2.9473	1.8694	.73660	.54997	.5126	-.00066	-.00500	-.0013	-.0100
8	2.9460	1.8594	.73570	.54930	.5124	-.00034	-.00468	-.0007	-.0094

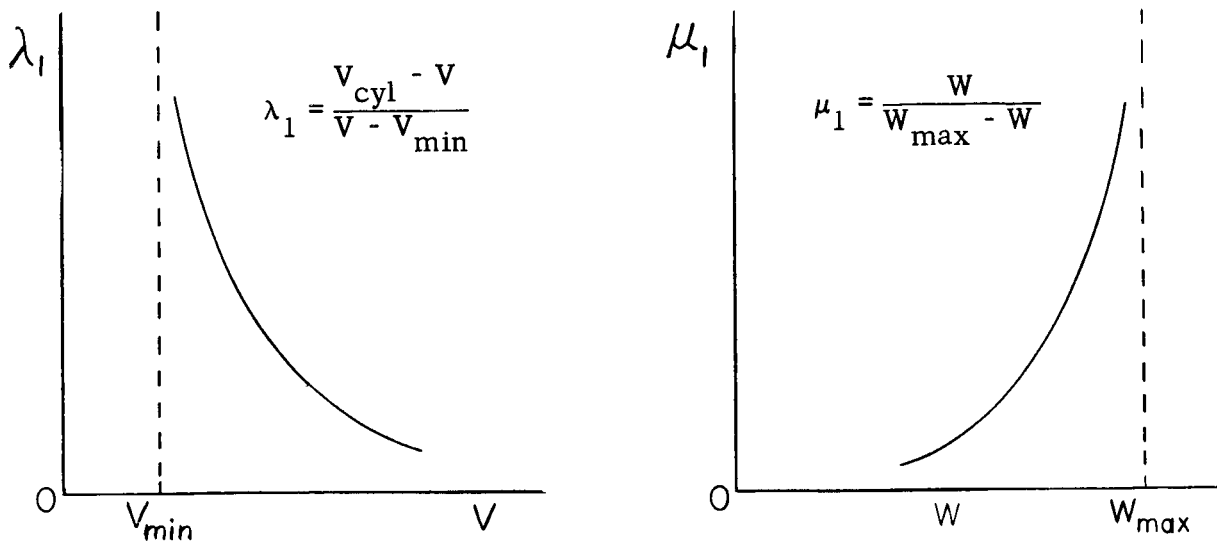


Fig. 1 Weighting Functions  $\lambda_1$  and  $\mu_1$

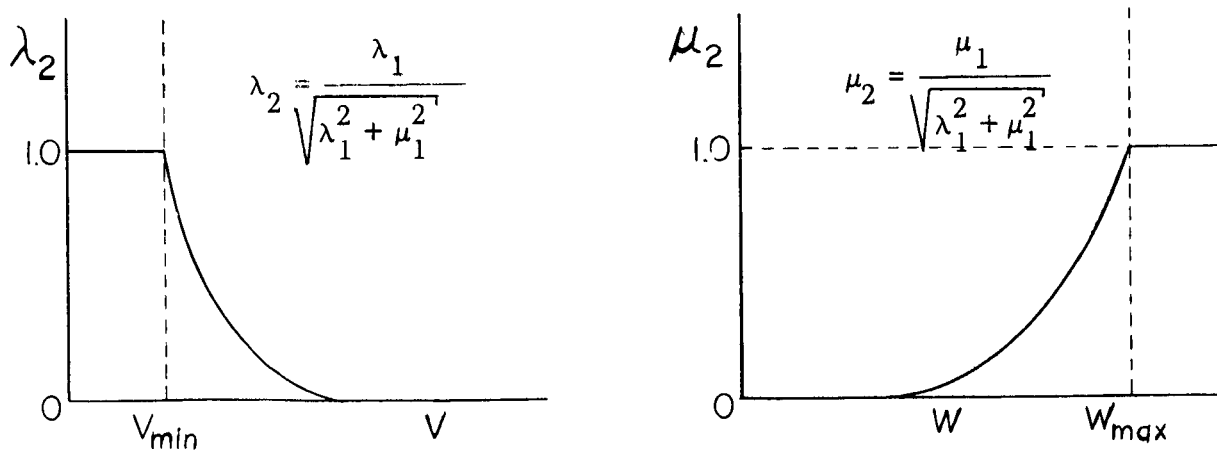


Fig. 2 Weighting Functions  $\lambda_2$  and  $\mu_2$



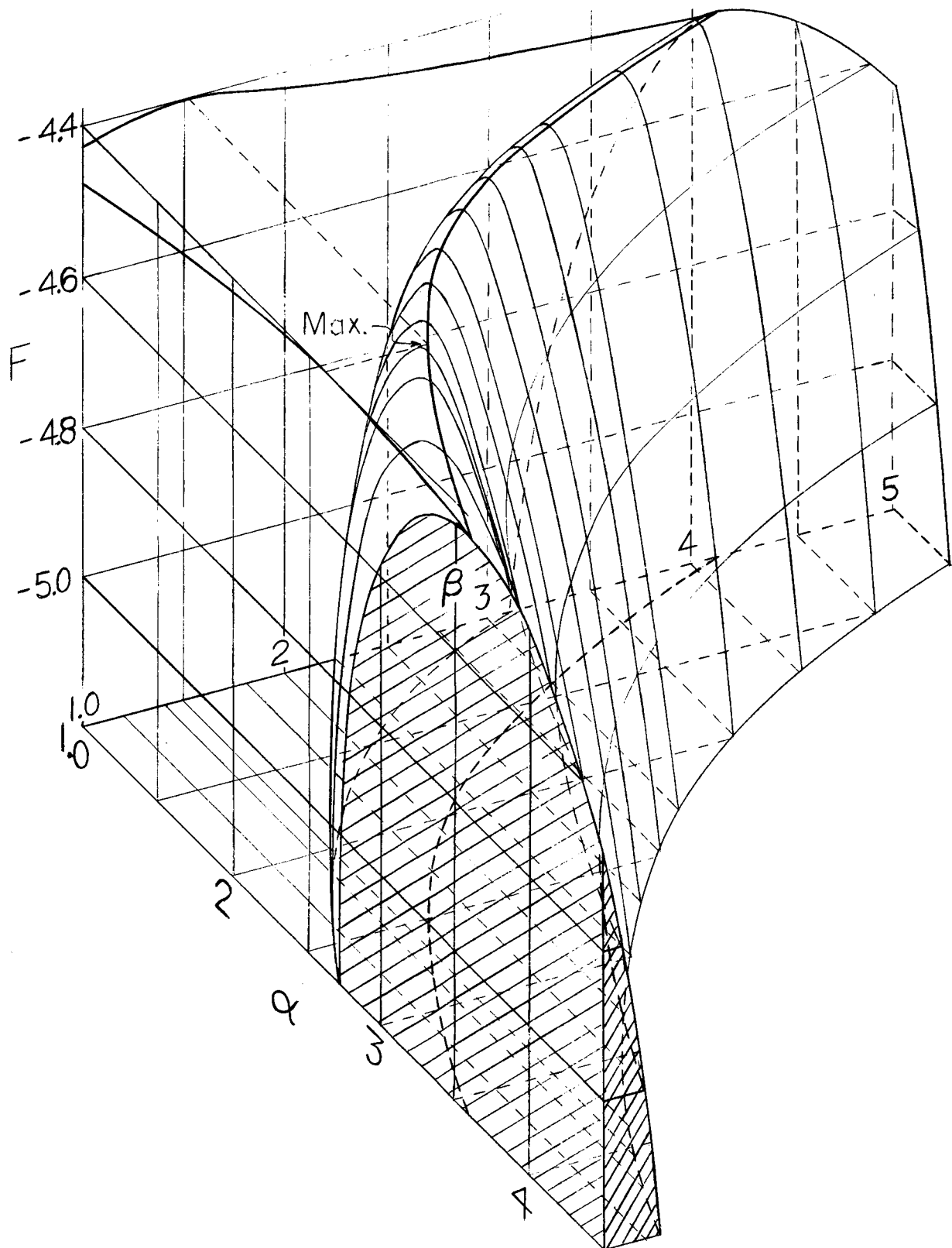


Fig. 3 Variation of the Function  $F$  with  $\alpha$  and  $\beta$

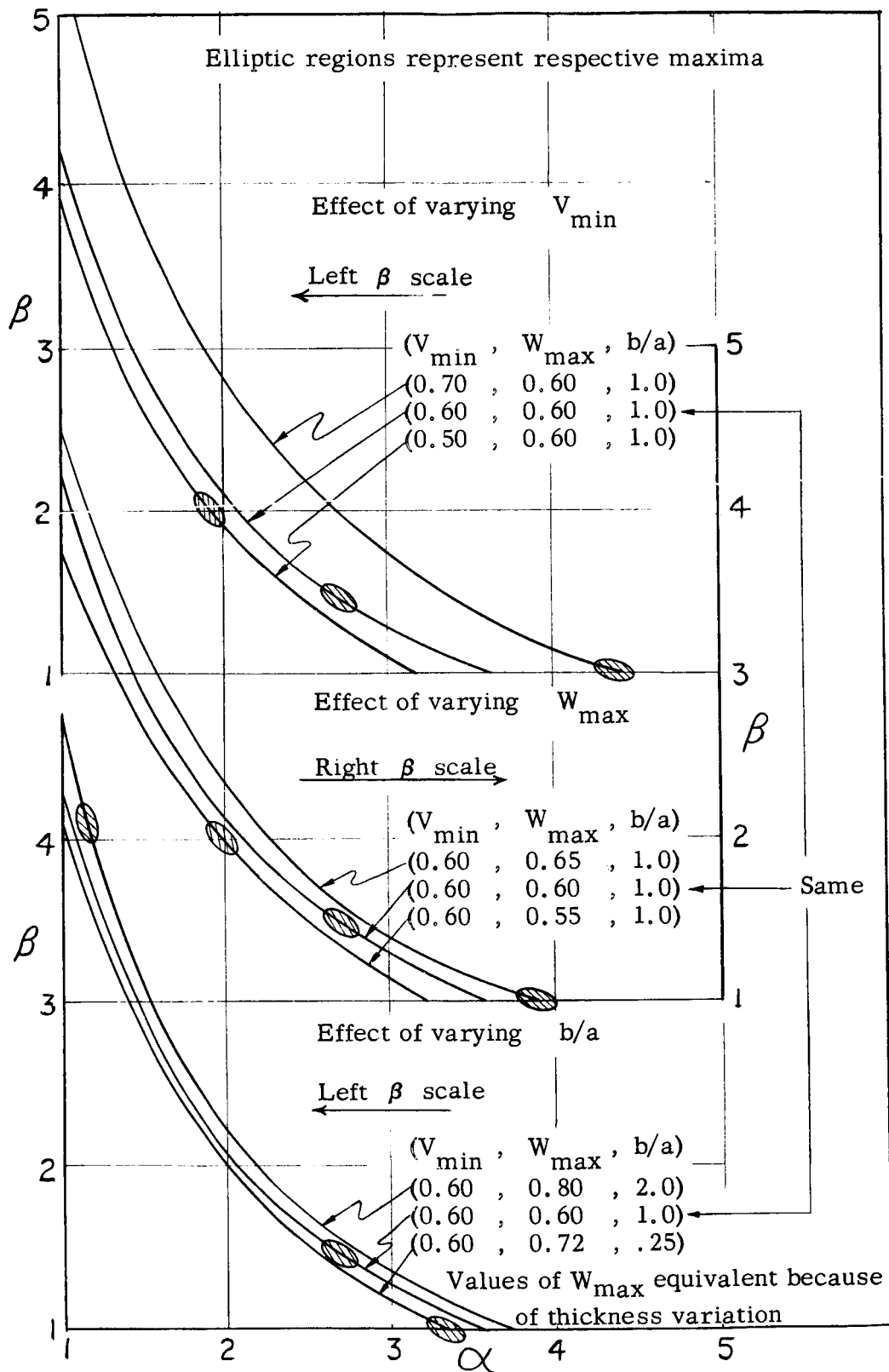


Fig. 4 Variation in Projections of Ridges on  $\alpha$ - $\beta$  Plane

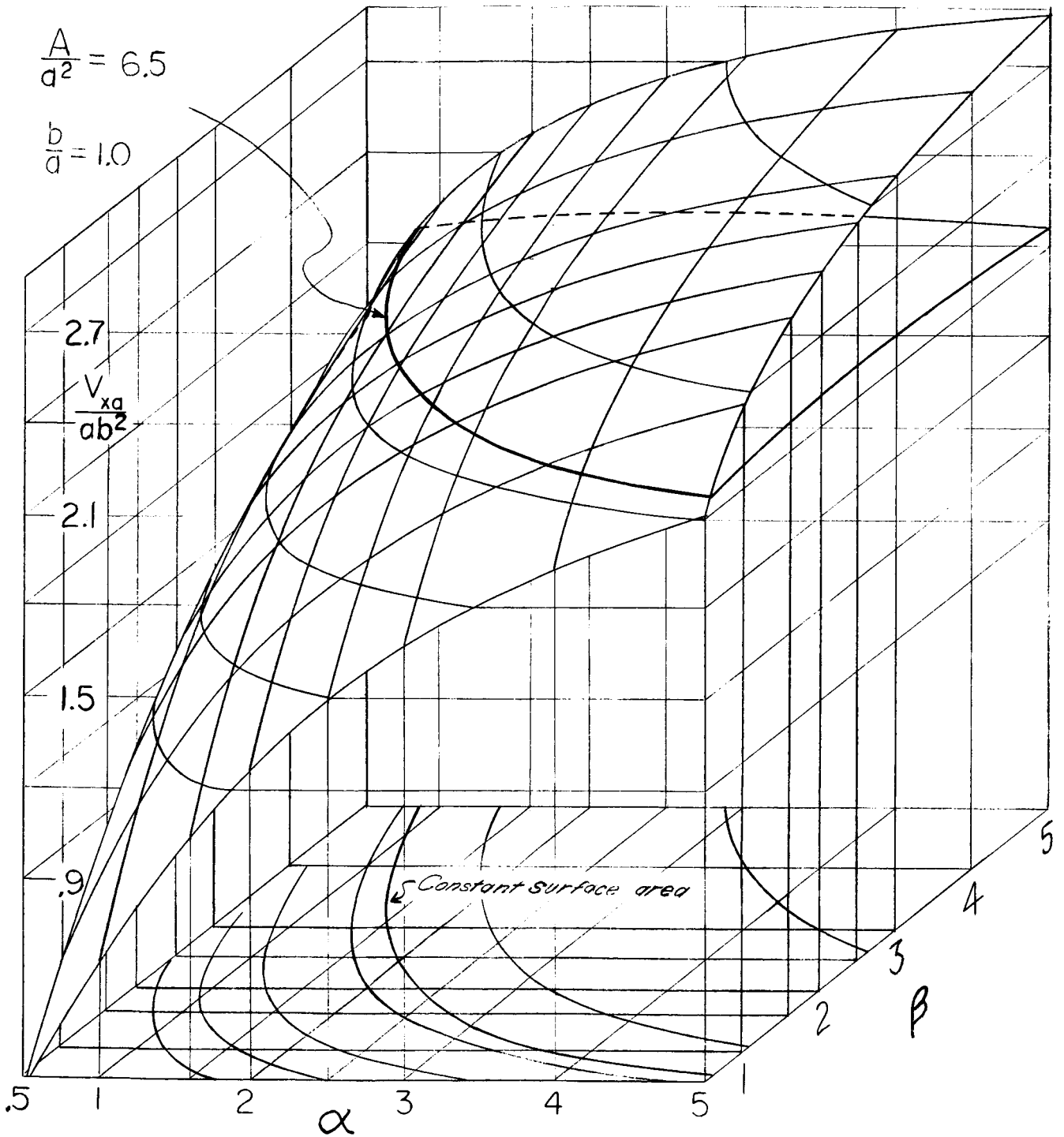


Fig.5 Variation in  $V_{xa}/ab^2$  with  $\alpha$  and  $\beta$

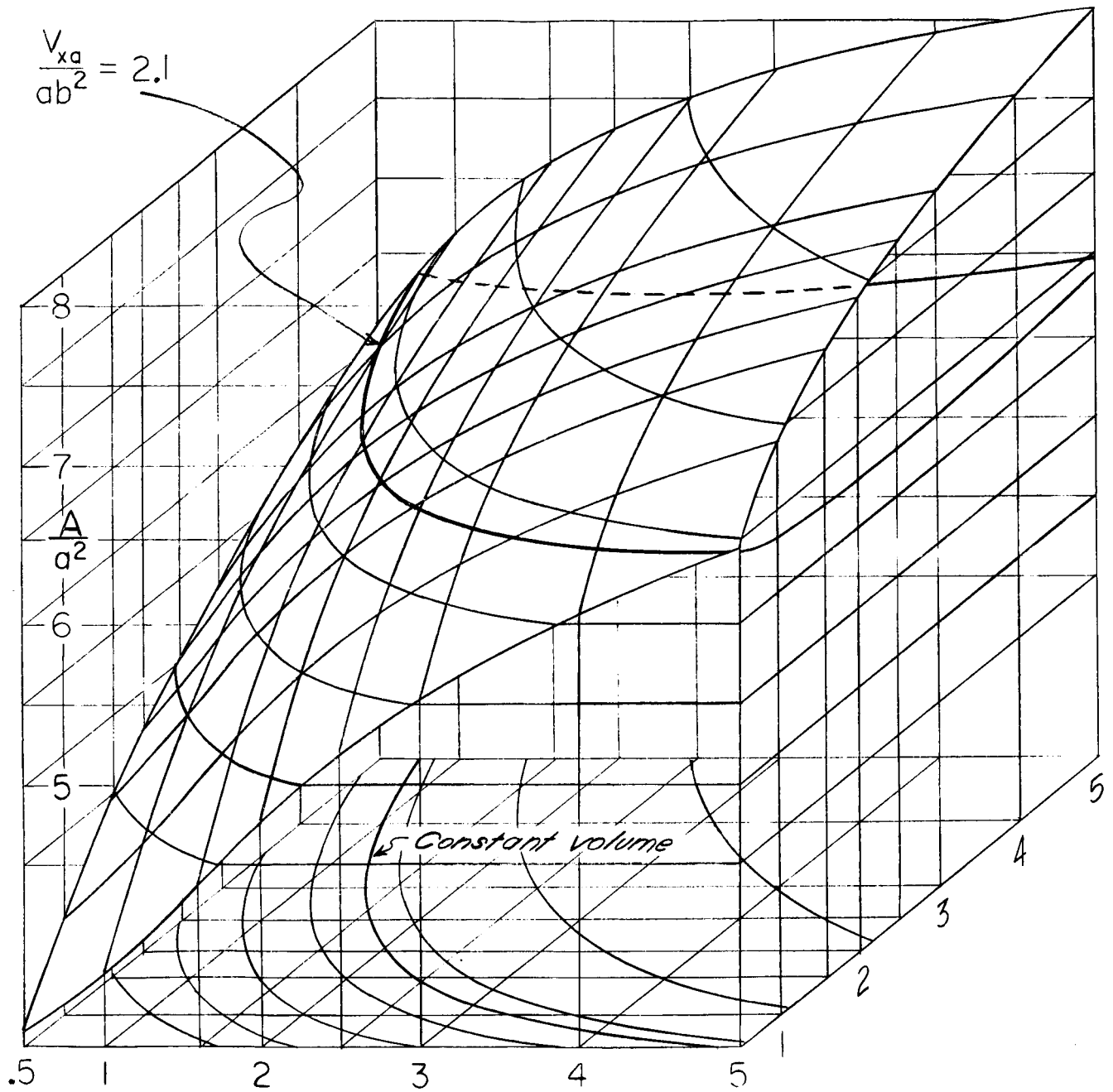


Fig.6 Variation in  $A/a^2$  with  $\alpha$  and  $\beta$  for  $b/a = 1.0$

Recent T. & A. M. Reports

<u>No.</u>	<u>Title</u>	<u>Date</u>
263	"Evaluation of Bonding Characteristics of Deformed Wire," by Russell S. Jensen and Clyde E. Kesler.	May, 1964
264	"Geometrical and Inertial Properties of a Class of Thin Shells of a General Type," by Will J. Worley and Han-chung Wang.	June, 1964
265	"The Fatigue Toughness of Metals: A Data Compilation," by Gary R. Halford.	June, 1964
266	"On a Theory for Axisymmetric Elastic Shells of Moderate Thickness," by R. J. Nikolai and A. P. Boresi.	July, 1964
267	"The Stress Distribution in a Notched Semi-infinite Plate," by D. Shadman.	August, 1964
268	"Yield Behavior of Niobium Single Crystals," by D. C. Huffaker.	September, 1964
269	"Euler Buckling of a Ring-Reinforced Cylindrical Shell Subjected to External Pressure," by H. L. Langhaar, A. P. Boresi and C. C. Fretwell.	September, 1964
270	"Photoelastic Study of the Stresses Near Openings in Pressure Vessels," by N. C. Lind and C. E. Taylor.	October, 1964
271	"Theoretical and Experimental Investigation of the Tensile Moduli of Parallel Filament Composites," by John W. Melvin.	November, 1964
272	"Third Conference on Fundamental Research in Plain Concrete," by Clyde E. Kesler.	November, 1964
273	"The Effect of Temperature on Cycle Dependent Deformation," by Brian R. Gain.	December, 1964
274	"An Investigation Into the Effect of Environmental Treatments on the Strength of E Glass Fibers," by N. M. Cameron.	January, 1965
275	"Crack Extension in Fiberglass Reinforced Plastics and a Critical Examination of the General Fracture Criterion," by E. M. Wu and R. C. Reuter, Jr.	January, 1965
276	"Applications of Lasers to Photoelasticity," by C. E. Taylor, C. E. Bowman, W. P. North, and W. F. Swinson.	February, 1965

