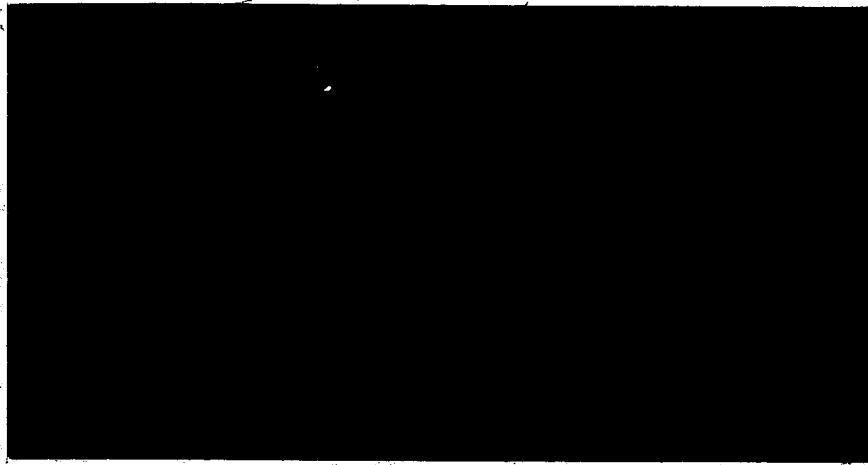




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CONICAL SHELL DEFLECTION
AND ACCELERATION RESPONSE
PROGRAM (DARC)

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SUBMITTED UNDER NASA Contract NAS 1-3179

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SUMMARY

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Detailed information is presented for using an IBM 7094 computer program which will calculate the response of truncated conical shells to random pressure excitation such as encountered by spacecraft adapters during a launch trajectory. This program was developed in connection with NASA (Langley Research Center) Contract NAS 1-3179.

A normal mode approach is used in the program formulation, with each mode represented by a second order linear oscillator. Options are provided for using either theoretical or experimental vibration modal data. The pressure excitation is handled by dividing the shell surface into many small areas and either (1) assuming a different homogeneous pressure over each resulting sub-area, or (2) using empirically determined pressure co-spectra to integrate the equations for each sub-area pair.

Author

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1. INTRODUCTION

This report is a program manual to be used in conjunction with an IBM 7094 computer program developed by McDonnell Aircraft Corporation for NASA (Langley Research Center) Contract NAS 1-3179.

The computer program will calculate the response of truncated conical shells to random pressure excitation induced by aerodynamic turbulence; a typical application would be the determination of spacecraft adapter response to buffeting pressures. The computed response functions are the normal coordinate deflection and acceleration power spectra, and the shell radial deflection and acceleration power spectra. The integrals of these functions with respect to frequency are also obtained which gives the associated mean-square responses.

The detailed response analysis upon which the program is based is given in References 1 and 2. In brief, this analysis consists of a normal mode approach with each mode represented by a second order linear oscillator. The exciting pressures are handled by dividing the shell surface into many small areas and either:

- (1) assuming a different homogeneous pressure over each resulting sub-area (constant correlation assumption).
- (2) using empirically determined pressure co-spectra and integrating the equations over each sub-area pair.

Either theoretical or experimental vibration modal data may be used in the computer program. The modes designated in this report as theoretical are derived in an analysis given in Reference 3 and briefly discussed in Reference 1. These are basically sine-cosine modes. A more general representation of the mode shapes is available in the form of a Fourier series. This later representation may be used for either theoretical or experimental modes.

The size of problem that can be solved (i.e., the number of normal modes and the number of areas into which the shell can be divided) is limited only by the core storage of the IBM 7094. At present the program can handle up to 25 normal modes and 112 areas.

Sections 3 and 4 give all necessary information for using the computer program. Section 3 is arranged such that the basic equations are covered first, with subsequent discussion of the variables appearing in the equations. Derivations associated with details of the program computations are also given in this section. Section 4 is slanted more toward actual program useage, i.e., input of data, program operations, deck set-up, etc.

A detailed table of contents is used herein for the purpose of separating the information of sections 3 and 4 into semi-outline form. This will enable the user to more readily obtain desired information.

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2. MAIN SYMBOLS

A	Sub-area or special matrix defined in text
A_m^n	Mode shape coefficient for displacement w
C	Power spectrum or co-spectrum depending upon alike or unlike superscripts
f	Frequency in cycles per second
L	Length of shell frustum
M	Generalized mass
m	Number of longitudinal half-waves of a given mode
n	Number of circumferential waves of a given mode
p	Pressure
q	Normal coordinate
R	Auto-correlation function
r	Radius of shell
x	Coordinate defining distance along cone generator from apex to a point on shell
t	Time
U_c	Convection velocity
w	Radial displacement
α	Streamwise decay rate
$\bar{\alpha}$	Complement of conical shell half-angle
β	Circumferential decay rate
ζ	Ratio of damping to critical damping
θ	Coordinate defining angular position on shell
λ_m	L/m
τ	Time shift for auto-correlation or cross-correlation functions

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Φ Radial mode shape
 Ω Reduced frequency
 ω Frequency in radians per second

Subscripts and Superscripts

A Indicates relationship to area
a Indicates response point a
i Indicates mode i
 k, l Indicate sub-areas k and l, respectively
N Indicates normalized function
p, q, w Indicate relationship to pressure, normal coordinate, and deflection, respectively

Matrix Notation

[] Square or Rectangular
[] Diagonal matrix
[] Row matrix
[]^T Transpose of matrix

NOTE: Double subscripts or superscripts appearing on symbols within a matrix imply all possible combinations of themselves as they vary from unity to their maximums. As the subscripts or superscripts vary, they give the row and column designation within the matrix; the first subscript or superscript gives the row, and the second gives the column. Thus, the variability of the subscripts or superscripts is limited by the number of rows and columns in the matrix.

Miscellaneous Notation

() Parenthesis following a symbol indicates that the symbol is a function of those variables appearing within the parenthesis
< > Indicates a time average

Differentiation of a variable with respect to time is indicated by dots over the variable.

3. ENGINEERING EXPLANATION OF PROGRAM OPERATIONS AND CAPABILITIES

3.1 Basic Equations. - As previously discussed, equations have been derived in Reference 1 for the response of a truncated conical shell to random pressure excitation. These equations result in the radial deflection and acceleration power spectrum of any point "a" on the shell surface. They are programmed on the IBM 7094 as follows:

Normal Coordinate Deflection Spectra

$$C_q^{ii}(f) = \frac{(1/M_1^2 \omega_1^4) [\Phi_{ik}^A] [C_p^{kl}(f)] [\Phi_{1l}^A]^T}{(1-(f/f_1)^2)^2 + (2\zeta_1(f/f_1))^2} \quad (1)$$

Normal Coordinate Acceleration Spectra

$$C_q^{ii}(f) = (\text{ACCEL.}) (2\pi)^4 f^4 C_q^{ii}(f) \quad (2)$$

NOTE: "ACCEL." is a constant which is input to the computer program in order that the units of C_q^{ii} be g^2/cps .

Radial Shell Deflection Spectra

$$C_w^{aa}(f) = \sum_i \Phi_{ai}^2 C_q^{ii}(f) \quad (3)$$

Radial Shell Acceleration Spectra

$$C_w^{aa}(f) = \sum_i \Phi_{ai}^2 C_q^{ii}(f) \quad (4)$$

Mean-Square Normal Coordinate Deflection

$$\langle q_i^2(t) \rangle = \int C_q^{ii}(f) df \quad (5)$$

Mean-Square Normal Coordinate Acceleration

$$\langle \ddot{q}_i^2(t) \rangle = \int C_q^{ii}(f) df = (\text{ACCEL.}) (2\pi)^4 \int f^4 C_q^{ii}(f) df \quad (6)$$

Mean-Square Radial Deflection

$$\langle \ddot{w}_a^2(t) \rangle = \int C_w^{aa}(f) df = \sum_i \Phi_{ai}^2 \langle \ddot{q}_i^2(t) \rangle \quad (7)$$

Mean-Square Radial Acceleration

$$\langle \ddot{w}_a^2(t) \rangle = \int C_w^{aa}(f) df = \sum_i \Phi_{ai}^2 \langle \ddot{q}_i^2(t) \rangle \quad (8)$$

The variables in Equations (1) through (8) are defined as follows:

- f Variable frequency in cycles/sec.
- f_i i th modal frequency in cycles/sec.
- ω_i i th modal frequency in radians/sec.
- M_i Generalized mass of i th mode
- ζ_i Ratio of damping to critical damping for mode i
- $\Phi_{ik}^A = \int_{A_k} \Phi(x, \theta) dA_k$ Integral over the k th area of the i th mode shape
- $[\Phi_{ik}^A]$ Row matrix formed as k varies from unity to the total number of areas
- $[C_p^{kl}(f)]$ Matrix of co-spectra of pressures formed as k and l take on all values from unity to the total number of areas. When $k = l$, $C_p^{kl}(f)$ is a power spectrum.
- Φ_{ai} Value of radial mode shape at point "a" in the i th mode

The determination of the variables needed to solve Equations (1) through (4) is discussed in subsequent sections; once these variables have been determined, the calculation of the deflection and acceleration spectra is relatively straightforward. Equations (5) and (6) are numerically evaluated by use of trapezoidal integration. Equations (7) and (8) are obtained from the results of Equations (5) and (6).

3.2 Co-Spectrum Matrix. - Information pertaining to the co-spectrum matrix is divided into the following five sub-sections.

3.2.1 Constant Correlation Assumption. - When utilizing the constant correlation assumption discussed in Reference 1, the pressure co-spectrum matrix, $[C_p^{kl}(f)]$, is computed from

$$[C_p^{kl}(f)] = \left[\sqrt{C_p^{kk}(f)} \right] [C_{pN}^{kl}(f)] \left[\sqrt{C_p^{kk}(f)} \right] \quad (9)$$

where $[C_{pN}^{kl}(f)]$ is a matrix of normalized co-spectra with its k th element given by

$$C_{pN}^{kl}(f) = C_p^{kl}(f) / \sqrt{C_p^{kk}(f)} \sqrt{C_p^{ll}(f)} \quad (10)$$

The input of the pressure power spectra, $C_p^{kk}(f)$, is relatively simple and is discussed in sections 3.2.3 and 4.3.10.3.

The normalized pressure co-spectrum information is input to the computer program in tabular form. Use of a straightforward approach would require the input of one table for each element of $[C_{pN}^{kl}(f)]$. With 112 the maximum order of the normalized co-spectrum matrix, this approach is impossible both from a human capability standpoint and from core storage limitations. To circumvent this problem, assumptions have been made which allow the computer program to form the normalized co-spectrum matrix from a limited amount of input data; i.e., the normalized co-spectrum matrix is expanded from a relatively small input matrix. Depending on the assumptions one wishes to make, the normalized co-spectrum matrix can be formed by either of two methods denoted Form One Expansion and Form Two Expansion, respectively.

The basic assumption involved with the Form One Expansion is that the normalized pressure co-spectra are independent of the shell coordinates (x and θ), but depend only on the increments Δx and $\Delta \theta$. The Form Two Expansion assumes that the normalized pressure co-spectra are independent of θ , but in contrast to Form One, they do depend upon longitudinal position x ; thus, the co-spectra depend upon x and $\Delta \theta$. The intricacies of these methods are covered in the explanations which follow. In these explanations, the subscript "p" in $[C_{pN}^{kl}(f)]$ can be dropped without loss of clarity if it is remembered that the discussions are concerned with pressure co-spectra only.

3.2.1.1 Form One Expansion - The Form One Expansion assumes that there exists circumferential correlation between three adjacent strips. Since the circumferential correlation is symmetric, right or left, only two strips (columns) are required for input, i.e., the correlation between areas 1 and 5 is identical to the correlation between areas 1 and 57. The number of sub-areas over which there is appreciable correlation longitudinally is indicated by the crosshatched section shown as an example case in Figure 3.2.1.1-1. Extension to the general case is straightforward. It should be noted that the size of crosshatched section is assumed to be independent of position on the frustum. In other words, the crosshatched section could have been shown anywhere on the frustum as long as it covers three areas longitudinally and two areas circumferentially. Since the normalized co-spectra are input only for the sub-areas in the crosshatched section, the input information will be explained as if areas 1, 2, 3, 5, 6 and 7 are included in the crosshatched section, as shown. It is required that the nor-

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Truncated Conical Shell with Form One Expansion

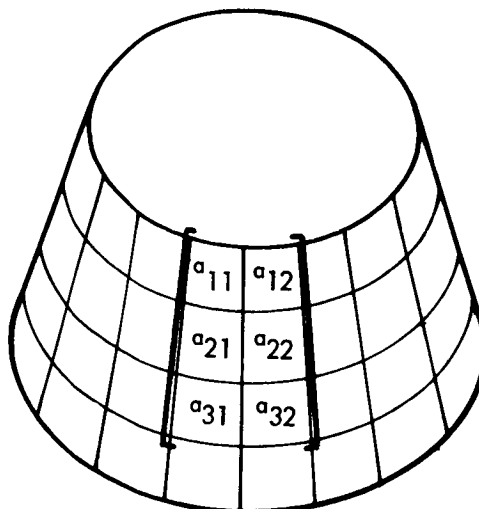
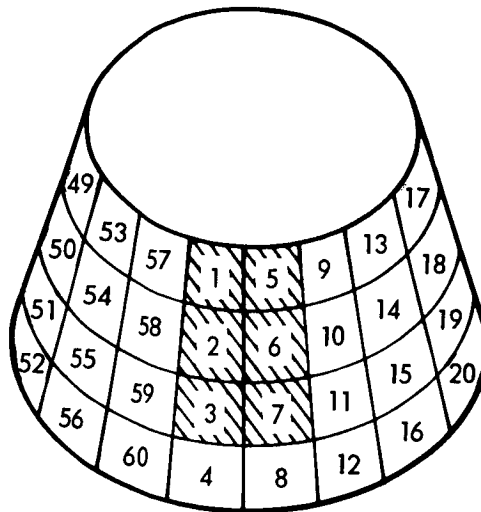


Figure 3.2.1.1-1

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malized co-spectrum between each area in the crosshatched section and the number 1 area be input as a co-spectra matrix defined as

$$[A] = \begin{bmatrix} a_{11} & a_{12} \\ a_{21} & a_{22} \\ a_{31} & a_{32} \end{bmatrix}$$

There is a specific relationship between the location of any element "a" with respect to the 1,1 element in [A] and the physical location of the corresponding area on the shell with respect to the number 1 area. For instance, a_{32} is two rows down and one column over from a_{11} . Therefore, it is the normalized co-spectrum between area 1 and the area two rows down and one column over, in this case area 7. The element a_{11} would be the normalized co-spectrum of area 1 with itself, which is equal to unity. In terms of the areas in the crosshatched section, the complete set of elements is as follows:

$$\begin{aligned} a_{11} &= C_N^{1,1} = 1.0 & a_{12} &= C_N^{1,5} \\ a_{21} &= C_N^{1,2} & a_{22} &= C_N^{1,6} \\ a_{31} &= C_N^{1,3} & a_{32} &= C_N^{1,7} \end{aligned}$$

The elements of [A] are read into the program by column.

To form the normalized co-spectra matrix for all areas, $[C_N^{kl}]$, the programming logic assumes that the normalized co-spectra depend only on relative position between two areas. Thus, C_N^{kl} is formed under the following assumptions:

$$C_N^{1,2} = C_N^{2,3} = C_N^{3,4} = C_N^{5,6} = C_N^{6,7} = \dots = C_N^{57,58} = C_N^{58,59} = C_N^{59,60}$$

$$C_N^{1,3} = C_N^{5,7} = \dots = C_N^{57,59}$$

$$C_N^{1,4} = C_N^{5,8} = \dots = C_N^{57,60}, \text{ where } C_N^{1,4} = 0 \text{ since, in this example, there is appreciable correlation over only three areas longitudinally.}$$

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$$C_N^{1,5} = C_N^{5,9} = \dots C_N^{53,57}, C_N^{1,57}$$

$$C_N^{1,5} = C_N^{2,6} = C_N^{6,10} = \dots C_N^{54,58}, C_N^{2,58}$$

$$\begin{array}{ccccccc} & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ C_N^{1,5} & = & C_N^{4,8} & = & C_N^{8,12} & = & \dots C_N^{56,60}, C_N^{4,60} \end{array}$$

$$C_N^{1,6} = C_N^{3,8} = C_N^{7,12} = C_N^{11,16}, \text{ etc.}$$

$$C_N^{1,6} = C_N^{1,58} = C_N^{57,54} = C_N^{53,50}, \text{ etc.}$$

$$C_N^{1,6} = C_N^{2,59} = C_N^{58,55} = C_N^{54,51}, \text{ etc.}$$

$$C_N^{1,6} = C_N^{3,60} = C_N^{59,56} = C_N^{55,52}, \text{ etc.}$$

$$C_N^{1,7} = C_N^{5,11} = C_N^{9,15} = C_N^{13,19}, \text{ etc.}$$

$$C_N^{1,7} = C_N^{2,8} = C_N^{6,12} = C_N^{10,16}, \text{ etc.}$$

$$C_N^{1,7} = C_N^{1,59} = C_N^{57,55} = C_N^{53,51}, \text{ etc.}$$

$$C_N^{1,7} = C_N^{2,60} = C_N^{58,56} = C_N^{54,62}, \text{ etc.}$$

Since, in this example, there is appreciable correlation over only three areas longitudinally and only two areas circumferentially, the program would use a zero co-spectrum between any areas separated either longitudinally by two areas or circumferentially by one area. For instance, $C_N^{5,8} = 0$ and $C_N^{2,10} = 0$. In forming $[C_N^{kl}]$ advantage is taken of the fact that $C_N^{kl} = C_N^{lk}$.

The size of the crosshatched section and the number of longitudinal areas are identified by the following input parameters:

- KL - number of longitudinal areas
- KC - number of longitudinal areas over which there is appreciable correlation
- KK - number of circumferential areas over which there is appreciable correlation

3.2.1.2 Form Two Expansion. - Figure 3.2.1.2-1 shows a truncated conical shell divided for illustrative purposes into four sections along x and fifteen sections circumferentially. The Form Two Expansion assumes that the correlation between areas in a longitudinal strip is the same for all strips, and correlation between any two areas not in the same strip is zero. Thus, if the co-spectra for the first four areas are input, the program can use this information for the other fourteen strips. The following matrix of co-spectra contains all necessary information for the first strip:

$$\begin{bmatrix} 1 & c_N^{1,2} & c_N^{1,3} & c_N^{1,4} \\ c_N^{1,2} & 1 & c_N^{2,3} & c_N^{2,4} \\ c_N^{1,3} & c_N^{2,3} & 1 & c_N^{3,4} \\ c_N^{1,4} & c_N^{2,4} & c_N^{3,4} & 1 \end{bmatrix} = [\Gamma] \quad (11)$$

The matrix of normalized co-spectra has been defined as $[\Gamma]$ for convenience. The superscripts correspond to the areas in the first strip. Since $[\Gamma]$ is symmetric and all diagonal terms equal unity, only the elements to the right of the diagonal are input. The order of input is by row, i.e.,

$$c_N^{1,2}, c_N^{1,3}, c_N^{1,4}, c_N^{2,3}, c_N^{2,4}, c_N^{3,4}.$$

Based on the assumptions of the Form Two Expansion, the total $[c_N^{kl}]$ is formed in terms of $[\Gamma]$ as follows:

$$[c_N^{kl}] = \begin{bmatrix} \textcircled{1} & & & & \\ [\Gamma] & \textcircled{2} & & & \\ & [\Gamma] & & & \\ & & \ddots & & \\ & & & \textcircled{15} & \\ & & & [\Gamma] & \end{bmatrix} \quad (12)$$

where the circled numbers indicate the longitudinal strips.

The number of longitudinal areas is given by KL. KC and KK have no meaning in Form Two.

3.2.2 Empirically Determined Correlation. - Empirically determined correlation is incorporated in the response equations by analytically describing the pressure co-spectra over the shell surface and subsequently performing the required integrations for each area and pair of areas into which the shell is subdivided. This approach is given detailed treatment in the appendix of Reference 1.

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Truncated Conical Shell with Form Two Expansion

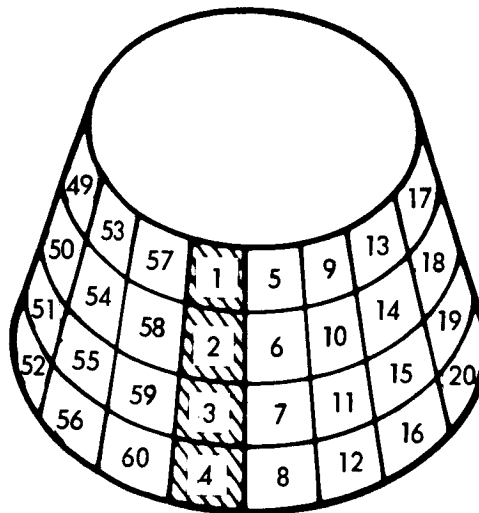


Figure 3.2.1.2-1

The effect of this approach on the computation scheme is to require the calculation of $C_N^k(f)$ for substitution into Equation (9) rather than inputting tables as is done with the constant correlation assumption method. As is the case with this latter method, however, the normalized co-spectra are not required for every pair of areas; i.e., all possible combinations of $C_N^k(f)$ need not be calculated. Instead, it is necessary to calculate only the elements of $[A]$ or $[\Gamma]$ which are required input when using the constant correlation assumption method. It is also possible to calculate specified elements of $[A]$ and $[\Gamma]$ and to use input tables for the unspecified elements.

The analytical expression used for the normalized co-spectra is based on the assumption of a convected decaying pressure field. This results in the following equation:

$$C_N(\xi, \eta, f) = e^{-\alpha\xi} e^{-\beta\eta} \cos \Omega \quad (13)$$

where:

$\xi = \Delta x$, streamwise separation distance

$\eta = r \Delta\theta$, circumferential separation

α = streamwise decay rate

β = circumferential decay rate

$\Omega = \frac{2\pi f \xi}{U_c}$, reduced frequency

U_c = streamwise convection velocity

The integration of Equation (13) for a particular pair of sub-areas results in

$$C_N^{kl}(f) = \frac{1}{[(\alpha\xi)^2 + \Omega^2](\beta\eta)^2} \left\{ [(\alpha\xi)^2 - \Omega^2] \cdot \right. \\ \left. [e^{-k\alpha\xi} \cos k \Omega + e^{-(k-2)\alpha\xi} \cos(k-2)\Omega - 2e^{-(k-1)\alpha\xi} \cos(k-1)\Omega] \right. \\ \left. + \delta_{k1} \alpha\xi \Omega (e^{\alpha\xi} - e^{-\alpha\xi}) \sin \Omega \right\} \left\{ e^{-l\beta\eta} + e^{-(l-2)\beta\eta} - 2e^{-(l-1)\beta\eta} \right\} \quad (14)$$

where

$$\delta_{k1} = 1 \quad \text{if } k = 1$$

$$\delta_{k1} = 0 \quad \text{if } k \neq 1$$

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As used in Equation (14), the superscripts k and l refer to area position longitudinally (streamwise) and circumferentially, respectively. Thus, they have the same meaning as the row and column designation identifying the elements of $[A]$. The relationship between the calculated normalized co-spectra and the elements of $[\Gamma]$ is not so straightforward, but this need not concern the reader since the relationship is handled by special programming.

3.2.2.1 Examples. - In order to clarify the above discussion, the examples covered in sections 3.2.1.1 and 3.2.1.2 will again be covered with specified normalized co-spectra to be calculated.

3.2.2.1.1 Form One Expansion. - The method for indicating that a particular normalized co-spectrum (particular element of $[A]$) is to be calculated consists of putting 21 on the input card which gives the number of points to be expected in the table following the aforementioned card. Actual tabular input must be accomplished with the number of points in the table less than or equal to 20 as indicated in sections 3.2.4 and 4.3.10.4.

Now for illustrative purposes, assume that the following elements of $[A]$ are to be calculated: a_{11} , a_{21} , a_{12} , a_{22} , a_{32} . Assume that it is desired to input element a_{31} as a table. The order and manner of input would be:

co-table ele. (1,1) - 21 points
co-table ele. (2,1) - 21 points
co-table ele. (3,1) - some specified no. of points less than 21
co-table ele. (1,2) - 21 points
co-table ele. (2,2) - 21 points
co-table ele. (3,2) - 21 points

3.2.2.1.2 Form Two Expansion. - Assume that the following required elements of $[\Gamma]$ are to be calculated: $C_N^{1,3}$, $C_N^{1,4}$, $C_N^{3,4}$. This leaves the elements $C_N^{1,2}$, $C_N^{2,3}$ and $C_N^{2,4}$ to be input by tables. The order and manner of input would be:

co-table ele. (1,2) - less than 21 points
co-table ele. (1,3) - 21 points
co-table ele. (1,4) - 21 points
co-table ele. (2,3) - less than 21 points
co-table ele. (2,4) - less than 21 points
co-table ele. (3,4) - 21 points

As previously mentioned, the computer program will automatically pick the correct k and l to be used in Equation (14); however, the input order must be as given above.

3.2.2.2 Required Input. - When using either of the methods discussed in sections 3.2.2.1.1 and 3.2.2.1.2, the following information with the appropriate units is required:

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<u>Input</u>	<u>Units</u>
α , streamwise decay rate	in. ⁻¹
β , circumferential decay rate	in. ⁻¹
Average size (length) of sub-area in longitudinal direction	in.
Average size (width) of sub-area in circumferential direction	in.
U_c , convection velocity	ft.sec. ⁻¹

In addition to the above, KL, KC, and KK are required and have the same meanings as when used with the constant correlation assumption methods.

3.2.3 Power Spectra Input. - Regardless of which form of normalized co-spectra input is utilized, there are three possible options concerning power spectra input. These options are the following:

- (1) Input only one power spectrum table, which is used for the complete shell.
- (2) Input a power spectrum table for each area in one strip into which the shell has been divided.
- (3) Input a power spectrum table for each area into which the shell has been divided.

The first option assumes a homogeneous distribution over the complete shell. The second option assumes circular symmetry for the pressure distribution; i.e., the input for a single strip is used for all strips. The third option allows for the power spectrum to vary from area-to-area circumferentially as well as longitudinally. This latter condition is representative of the situation encountered by spacecraft adapters for non-zero angle of attack. When it is used, one is, in effect, assuming that only the magnitude of the exciting pressure is changed by angle of attack, with the correlation (normalized co-spectra) remaining unaffected. This assumption is shown to be fairly reasonable by the data presented in References 4 and 5.

3.2.4 Spectra Tables. - The spectra tables are restricted to twenty points each. The first frequency in the table must be equal to or less than the smallest calculating frequency. The largest calculating frequency must be exceeded or equalled by the largest frequency in the table. The frequencies in each table are independent of the frequencies in any other table. Thus, different sets of frequencies may be used for each table if desired.

When scaled model wind tunnel tests provide the spectral data, as is usually the case, it is necessary to multiply the model frequencies and power spectrum levels by appropriate constants in order to convert the data to full scale. As a convenience, provisions have been made in the computer program to multiply the frequencies of all tables (both power spectra and normalized co-spectra) and the power spectrum levels by input constants. This is not an option, i.e., constants must always be input. Thus, if it is not desired to scale the frequencies and power spectrum levels, the constants must be input as unity.

3.2.5 Mathematical Definition of Power Spectra. - The power spectrum levels are defined such that the area under a power spectrum curve gives the mean-square value. Mathematically,

$$C_p^{kk}(f) = 4 \int_0^{\infty} \cos(2\pi f\tau) R_p^{kk}(\tau) d\tau$$

where $R_p^{kk}(\tau)$ is an auto-correlation function.

3.3 Calculation of $[\Phi^A]^T$. - The row matrices $[\Phi_{ik}^A]$ and $[\Phi_{il}^A]$ appearing in Equation (1) make up the rows of a matrix defined as $[\Phi^A]^T$. That is,

$$[\Phi^A]^T = \begin{bmatrix} [\Phi_{1k}^A] \\ [\Phi_{2k}^A] \\ \vdots \\ [\Phi_{ik}^A] \end{bmatrix}$$

Note that k and l can be used interchangeably since, when used inside a matrix, each varies from one to the total number of areas into which the shell has been subdivided. The k th element of $[\Phi_{ik}^A]$ is the integral over the k th area of the i th mode shape and can be written as

$$\Phi_{ik}^A = \int_{A_k} \Phi_i(x, \theta) dA_k \quad (15)$$

where $\Phi_i(x, \theta)$ is the i th mode radial deflection of a truncated conical shell.

3.3.1 Theoretical Modes. - The mode shape $\Phi_i(x, \theta)$ is given in Reference 1 for a theoretical set of modes and has the form

$$\Phi_i(x, \theta) = A_{m_1}^n x \cos(\bar{\alpha}) \sin \frac{x-x_1}{\lambda_{m_1}} \cos(n_1 \theta) \quad (16)$$

The variables in Equation (16) are defined as follows:

- $A_{m_i}^n$ - Amplitude factor of i th mode
- x - As shown in Figure 3.3.1-1
- $\bar{\alpha}$ - As shown in Figure 3.3.1-1
- x_1 - As shown in Figure 3.3.1-1
- λ_{m_i} - $L/m_i\pi$
- L - Length of frustum, $x_E - x_1$
- m_i - Number of longitudinal half-waves
- n_i - Number of circumferential waves

In order to determine Φ_{ik}^A , Equation (16) is used to substitute for $\Phi_i(x, \theta)$ in Equation (15) and A_k is put in a more definitive form. This results in

$$\Phi_{ik}^A = A_{m_i}^n \cos(\bar{\alpha}) \int_{x_u}^{x_{u+1}} \int_{\theta_v}^{\theta_{v+1}} x \sin\left(\frac{x-x_1}{\lambda_{m_i}}\right) \cos(n_i \theta) r(x) d\theta dx \quad (17)$$

The variables x_u , x_{u+1} , θ_v , θ_{v+1} define the limits associated with A_k (see Figure 3.3.1-1). Substitution of $r(x) = x \cos(\bar{\alpha})$ in Equation (17) gives

$$\Phi_{ik}^A = A_{m_i}^n \cos^2(\bar{\alpha}) \int_{x_u}^{x_{u+1}} \int_{\theta_v}^{\theta_{v+1}} x^2 \sin\left(\frac{x-x_1}{\lambda_{m_i}}\right) \cos(n_i \theta) d\theta dx \quad (18)$$

The integral of Equation (18) is given by

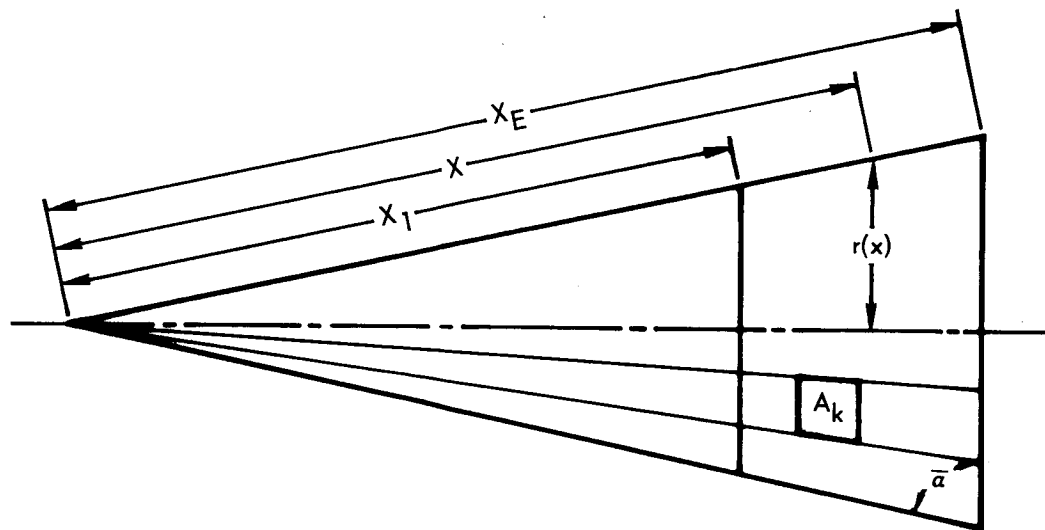
$$\begin{aligned} \Phi_{ik}^A = & \bar{\beta} \lambda_{m_i}^3 \cos\left(\frac{x_1}{\lambda_{m_i}}\right) \left\{ - \left(\frac{x_{u+1}^2}{\lambda_{m_i}^2} \cos\left(\frac{x_{u+1}}{\lambda_{m_i}}\right) - \frac{x_u^2}{\lambda_{m_i}^2} \cos\left(\frac{x_u}{\lambda_{m_i}}\right) \right) \right. \\ & + 2 \left(\frac{x_{u+1}}{\lambda_{m_i}} \sin\left(\frac{x_{u+1}}{\lambda_{m_i}}\right) - \frac{x_u}{\lambda_{m_i}} \sin\left(\frac{x_u}{\lambda_{m_i}}\right) \right) + 2 \left(\cos\left(\frac{x_{u+1}}{\lambda_{m_i}}\right) - \cos\left(\frac{x_u}{\lambda_{m_i}}\right) \right) \left. \right\} \\ & - \bar{\beta} \lambda_{m_i}^3 \sin\left(\frac{x_1}{\lambda_{m_i}}\right) \left\{ \left(\frac{x_{u+1}^2}{\lambda_{m_i}^2} \sin\left(\frac{x_{u+1}}{\lambda_{m_i}}\right) - \frac{x_u^2}{\lambda_{m_i}^2} \sin\left(\frac{x_u}{\lambda_{m_i}}\right) \right) \right. \\ & + 2 \left(\frac{x_{u+1}}{\lambda_{m_i}} \cos\left(\frac{x_{u+1}}{\lambda_{m_i}}\right) - \frac{x_u}{\lambda_{m_i}} \cos\left(\frac{x_u}{\lambda_{m_i}}\right) \right) - 2 \left(\sin\left(\frac{x_{u+1}}{\lambda_{m_i}}\right) - \sin\left(\frac{x_u}{\lambda_{m_i}}\right) \right) \left. \right\} \quad (19) \end{aligned}$$

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Truncated Conical Shell Theoretical Modal Representation



Section of Frustum

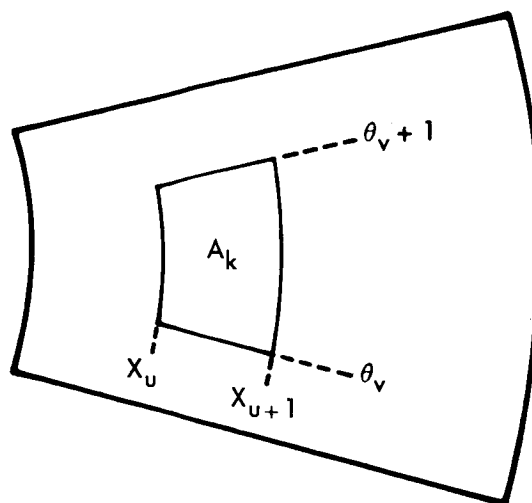


Figure 3.3.1-1

where:

$$\bar{\rho} = \frac{1}{n_i} \left(A_{m_i}^n \cos^2(\bar{\alpha}) \right) \left(\sin(n_i \theta_{v+1}) - \sin(n_i \theta_v) \right), n_i \neq 0$$

$$\bar{\rho} = \left(A_{m_i}^n \cos^2(\bar{\alpha}) \right) (\theta_{v+1} - \theta_v), n_i = 0 \quad (20)$$

Equations (19) and (20) are solved for each area into which the shell is subdivided. The order in which the areas are covered follows a specific sequence. The areas are numbered as shown in Figures 3.2.1.1-1 and 3.2.1.2-1. The area numbers correspond to k in $[\phi_{ik}^A]$ which is the ith row of $[\phi^A]^T$. The example configuration shown in Figures 3.2.1.1-1 and 3.2.1.2-1 is for four equal divisions along x and fifteen equal divisions circumferentially; however, the user would specify the number of divisions for an actual problem, subject to the limitations given in section 4.3.6.1.

3.3.2 Fourier Series Modes. - When using a Fourier series modal representation, $[\phi^A]^T$ is formed in a manner similar to that used with theoretical modes, except a double Fourier series instead of Equation (16) is used for $\phi_1(x, \theta)$. Referring to Figure 3.3.2-1, the double Fourier series in terms of $\bar{x}$ and θ is

$$\phi_1(\bar{x}, \theta) = \left\{ \frac{a_{0i}}{2} + \sum_m \left(a_{m_i} \cos\left(\frac{m_i \pi}{\bar{L}} \bar{x}\right) + b_{m_i} \sin\left(\frac{m_i \pi}{\bar{L}} \bar{x}\right) \right) \right\} \left\{ \frac{c_{0i}}{2} + \sum_n \left(c_{n_i} \cos(n_i \theta) + d_{n_i} \sin(n_i \theta) \right) \right\} \quad (21)$$

where,

- x - Coordinate defining longitudinal distance along part of shell to be analyzed
- L - Length of shell frustum to be analyzed
- $\bar{x}$ - Coordinate to which Fourier series is referenced
- $\bar{L}$ - Length over which Fourier series is defined
- γ - Distance between $\bar{x}$ and $x(\bar{x} - x = \gamma)$, positive as shown in Figure 3.3.2-1
- m_i - Number of longitudinal half-waves
- n_i - Number of circumferential waves
- a_{m_i} - Coefficient for cosine terms of longitudinal series

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Truncated Conical Shell Fourier Series Modal Representation

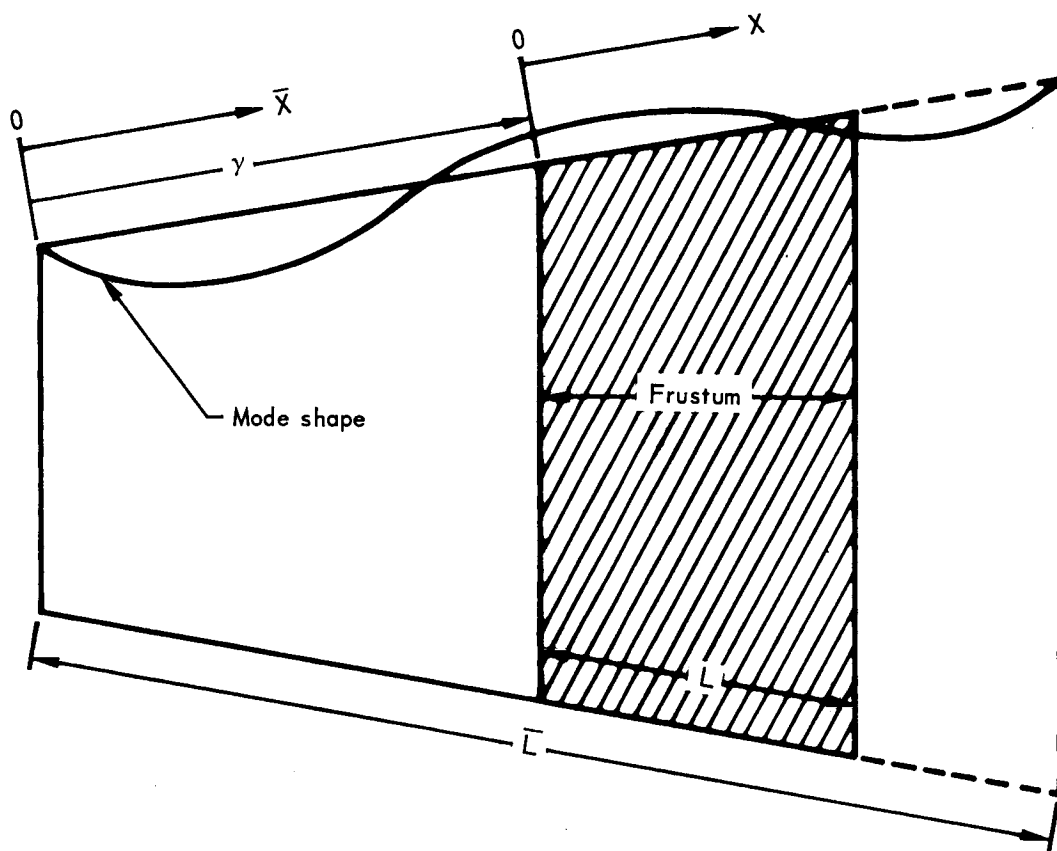


Figure 3.3.2-1

b_{m1} - Coefficient for sine terms of longitudinal series

c_{n1} - Coefficient for cosine terms of circumferential series

d_{n1} - Coefficient for sine terms of circumferential series

Substituting $\bar{x} = x + \gamma$ into Equation (21) and simplifying gives

$$\begin{aligned} \Phi_1(s, \theta) = & \left\{ \frac{a_{01}}{2} + \sum_m \left[a_{m1} \left(\cos\left(\frac{m_1\pi}{L} x\right) \cos\left(\frac{m_1\pi}{L} \gamma\right) - \sin\left(\frac{m_1\pi}{L} x\right) \sin\left(\frac{m_1\pi}{L} \gamma\right) \right) \right. \right. \\ & \left. \left. + b_{m1} \left(\sin\left(\frac{m_1\pi}{L} x\right) \cos\left(\frac{m_1\pi}{L} \gamma\right) + \cos\left(\frac{m_1\pi}{L} x\right) \sin\left(\frac{m_1\pi}{L} \gamma\right) \right) \right] \right\} \cdot \\ & \left\{ \frac{c_{01}}{2} + \sum_n \left[c_{n1} \cos(n_1\theta) + d_{n1} \sin(n_1\theta) \right] \right\} \end{aligned} \quad (22)$$

Writing Equation (22) in a simpler manner and substituting into Equation (15) results in

$$\begin{aligned} \Phi_{1k}^A = & \int_{x_u}^{x_{u+1}} \int_{\theta_v}^{\theta_{v+1}} \left\{ \left[\frac{a_{01}}{2} + \sum_m \left(\bar{a}_{m1} \cos\left(\frac{m_1\pi}{L} x\right) + \bar{b}_{m1} \sin\left(\frac{m_1\pi}{L} x\right) \right) \right] \left[\frac{c_{01}}{2} + \right. \right. \\ & \left. \left. \sum_n \left(c_{n1} \cos(n_1\theta) + d_{n1} \sin(n_1\theta) \right) \right] r(x) d\theta dx \right\} \end{aligned} \quad (23)$$

where

$$\bar{a}_{m1} = a_{m1} \cos\left(\frac{m_1\pi}{L} \gamma\right) + b_{m1} \sin\left(\frac{m_1\pi}{L} \gamma\right) \quad (24)$$

$$\bar{b}_{m1} = -a_{m1} \sin\left(\frac{m_1\pi}{L} \gamma\right) + b_{m1} \cos\left(\frac{m_1\pi}{L} \gamma\right) \quad (25)$$

Assuming that $r(x)$ can be approximated satisfactorily by the average radius of the crosshatched section, r_{av} , the integration indicated in Equation (23) gives the following result:

$$\begin{aligned} \Phi_{1k}^A = & \frac{r_{av} \bar{L}}{\pi} \left\{ \frac{a_{01}\pi}{2\bar{L}} (x_{u+1} - x_u) + \sum_m \frac{1}{m_1} \left[\bar{a}_{m1} \left(\sin\left(\frac{m_1\pi}{L} x_{u+1}\right) - \sin\left(\frac{m_1\pi}{L} x_u\right) \right) \right. \right. \\ & \left. \left. - \bar{b}_{m1} \left(\cos\left(\frac{m_1\pi}{L} x_{u+1}\right) - \cos\left(\frac{m_1\pi}{L} x_u\right) \right) \right] \right\} \left\{ \frac{c_{01}}{2} (\theta_{v+1} - \theta_v) \right. \\ & \left. + \sum_n \frac{1}{n_1} \left[c_{n1} \left(\sin(n_1\theta_{v+1}) - \sin(n_1\theta_v) \right) - d_{n1} \left(\cos(n_1\theta_{v+1}) - \cos(n_1\theta_v) \right) \right] \right\} \end{aligned} \quad (26)$$

Assuming that the section of frustum to be analyzed is divided into equal segments longitudinally and circumferentially, i.e., $x_{u+1} - x_u = \Delta x$ and $\theta_{v+1} - \theta_v = \Delta\theta$, Equation (26) can be written as

$$\begin{aligned} \Phi_{ik}^A = \frac{r_{av} \bar{L}}{\pi} \left\{ \frac{a_{o1} \pi}{\bar{L}} \Delta x + \sum_m \frac{1}{m_1} \left[\bar{a}_{m1} \left(\mu_{m1} \sin \left(\frac{m_1 \pi}{\bar{L}} x_u \right) + \nu_{m1} \cos \left(\frac{m_1 \pi}{\bar{L}} x_u \right) \right) \right. \right. \\ \left. \left. - \bar{b}_{m1} \left(\mu_{m1} \cos \left(\frac{m_1 \pi}{\bar{L}} x_u \right) - \nu_{m1} \sin \left(\frac{m_1 \pi}{\bar{L}} x_u \right) \right) \right] \right\} \left\{ \frac{c_{o1}}{2} \Delta\theta + \right. \\ \left. \sum_n \frac{1}{n_1} \left[c_{n1} \left(\mu_{n1} \sin(n_1 \theta_v) + \nu_{n1} \cos(n_1 \theta_v) \right) \right. \right. \\ \left. \left. - d_{n1} \left(\mu_{n1} \cos(n_1 \theta_v) - \nu_{n1} \sin(n_1 \theta_v) \right) \right] \right\} \end{aligned} \quad (27)$$

where

$$\mu_{m1} = \left[\cos \left(\frac{m_1 \pi}{\bar{L}} \Delta x \right) \right] - 1 \quad (28)$$

$$\nu_{m1} = \sin \left(\frac{m_1 \pi}{\bar{L}} \Delta x \right) \quad (29)$$

$$\mu_{n1} = \left[\cos(n_1 \Delta\theta) \right] - 1 \quad (30)$$

$$\nu_{n1} = \sin(n_1 \Delta\theta) \quad (31)$$

Equations (24), (25), and (27) through (31) are solved for each area into which the shell has been divided. The order in which the areas are covered follows the same sequence as given for the theoretical modes.

3.4 Deflection and Acceleration Calculations - From Equation (3) of section 3.1, it can be seen that the response deflection calculation requires Φ_{ai}^2 for all combinations of a and i .

3.4.1 Theoretical Modes. - It will be shown in this section how ϕ_{ai}^2 is related to the theoretical modes given in Reference 1. The equation for the i th radial mode, as written in Reference 1, is

$$\phi_i(x, \theta)_c = A_{m_i}^n x \cos(\bar{\alpha}) \sin\left(\frac{x-x_1}{\lambda_{m_i}}\right) \cos(n_i \theta) \quad (32)$$

The subscript "c" indicates cosine mode. To obtain a completely general solution, a sine mode in θ must also be included in the analysis. The equation for this mode is

$$\phi_i(x, \theta)_s = A_{m_i}^n x \cos(\bar{\alpha}) \sin\left(\frac{x-x_1}{\lambda_{m_i}}\right) \sin(n_i \theta) \quad (33)$$

With both these modes included in the analysis, the deflection response is given as

$$C_w^{aa}(f) = \sum_i \phi_i^2(x_a, \theta_a)_c C_{qc}^{ii}(f) + \phi_i^2(x_a, \theta_a)_s C_{qs}^{ii}(f) \quad (34)$$

where $C_{qc}^{ii}(f)$ and $C_{qs}^{ii}(f)$ are the normal coordinate power spectra for the i th cosine mode and i th sine mode, respectively.

In order to write Equation (34), it is necessary to neglect all cross terms between the normal coordinate responses (normal coordinate co-spectra) in the same manner as is done in Reference 1. Computer studies have indicated this to be a valid assumption, even for the co-spectrum between a sine and cosine pair of modes having the same modal frequencies. These co-spectra have been found to be several orders of magnitude less than the associated normal coordinate power spectra, C_{qc}^{ii} and C_{qs}^{ii} . A further result of the above computer studies is that, for all practical purposes, C_{qc}^{ii} and C_{qs}^{ii} are equal. These findings are for a symmetric pressure excitation, however, and are not valid for non-symmetric excitation.

Equation (34) is simplified and the amount of computation substantially reduced if it is assumed that $C_{qc}^{ii} = C_{qs}^{ii}$. Thus, Equation (34) is written

$$C_w^{aa}(f) = \sum_i \left(\phi_i^2(x_a, \theta_a)_c + \phi_i^2(x_a, \theta_a)_s \right) C_{qc}^{ii}(f) \quad (35)$$

By setting

$$\left(\phi_i^2(x_a, \theta_a)_c + \phi_i^2(x_a, \theta_a)_s \right) = \phi_{ai}^2$$

Equation (35) becomes Equation (3). Substituting Equations (32) and (33) into the above expression for ϕ_{ai}^2 results in

$$\phi_{ai}^2 = \left(A_{m_i}^n x_a \cos(\bar{\alpha}) \right)^2 \sin^2 \left(\frac{x_a - x_1}{\lambda_{m_i}} \right) \quad (36)$$

The above equation is used to determine the required ϕ_{ai}^2 . The values of x at which the responses are desired are input to the computer program. The calculated deflection and acceleration response spectra are printed out in the same order as the x 's are read in.

It should be noted that the assumption used to write Equation (35), i.e., $C_{qs}^{ii} = C_{qc}^{ii}$, cannot be used with the third power spectrum input option covered in section 3.2.3.

3.4.2 Fourier Series Modes. - When using a Fourier series modal representation, the values of ϕ_{ai}^2 are read directly into the computer program. They are stored in a matrix designated $[\phi^2]^T$ and are arranged in this matrix as follows:

$$[\phi^2]^T = \begin{bmatrix} \phi_{11}^2 & \phi_{21}^2 & \dots & \phi_{a1}^2 \\ \phi_{12}^2 & \phi_{22}^2 & \dots & \phi_{a2}^2 \\ \vdots & \vdots & & \vdots \\ \phi_{1i}^2 & \phi_{2i}^2 & \dots & \phi_{ai}^2 \end{bmatrix}$$

Thus, each row of the above matrix gives the modal deflections squared of a particular mode.

As is the case with theoretical modes, the response spectra are printed out in the same order as the ϕ_{ai}^2 are input.

3.5 Input of Generalized Masses, Modal Damping, and Modal Frequencies. - The generalized masses, modal damping ratios, and modal frequencies are input in the same order as the rows of $[\phi^A]^T$. In other words, the i th input generalized mass, damping ratio, and frequency are related to the i th row of $[\phi^A]^T$.

3.6 Calculation Frequencies. - The calculating frequencies are best explained by an example. Let a set of calculating frequencies be given by the following:

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<u>Calculation Frequency</u>	<u>Calculation Interval</u>	<u>No. of Calculations ÷ No. of Printouts</u>
50.	2.	1.
60.	3.	1.
70.	1.	2.
81.		

Between 50 and 60 cps, the program will calculate every 2 cps and printout each time it calculates. Between 60 and 70 cps, the program will calculate and printout 63, 66, 69 and 70 cps. Between 70 and 81 cps, the program will calculate every 1 cps and printout at 72, 74, 76, 68, 80 and 81 cps.

In addition to the above frequencies, the program also calculates and prints out at each of the modal frequencies.

3.7 Units. - In addition to the specific units required in section 3.2.2, the modal frequencies f_1 must be input with units of cycles/sec. Also when using theoretical modes, α is always input in degrees.

Any consistent set of units may be used for the other input variables; a possible set is as follows:

<u>Input</u>	<u>Units</u>
$C^{kk}(f)$	$(lb/in^2)^2/cps$
$[\phi^A]^T$	in^2
$[\phi^2]^T$	Dimensionless
M_1	$lb \cdot sec^2/in.$

<u>Output</u>	<u>Units</u>
Normal Coordinate Deflection Spectra	in^2/cps
Normal Coordinate Acceleration Spectra	$(in/sec^2)^2/cps$ with ACCEL. = 1.0 g^2/cps with ACCEL. = 1/386
Radial Shell Deflection Spectra	in^2/cps
Radial Shell Acceleration	$(in/sec^2)^2/cps$ with ACCEL. = 1.0 g^2/cps with ACCEL. = 1/386

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There may be some confusion about the elements of $[\phi^2]^T$ having no units. This is because of the x in Equations (32) and (33), which apparently gives ϕ_{ai}^2 units of inches². It should be recognized, however, that the x , as used in these equations, is a mode shape modifying factor only. Another possible modifying factor, which would not cause confusion, would be e^{hx} . This latter factor is obviously dimensionless, although it does depend on x .

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4. COMPUTATIONAL PROCEDURES AND DETAILED PROGRAM INFORMATION

4.1 General Program Discussion. - The Deflection and Acceleration Response Calculation Digital Computer Program (DARC) will calculate the random response of the frustum of a moderately tapered conical shell, or section of a cylinder, subjected to random pressure excitation.

4.1.1 Language. - This program was originally written in FORTRAN II and later converted to FORTRAN IV. This write-up pertains only to the FORTRAN IV version.

4.1.2 Operations. - The program requires only standard input-output tapes with no special setup. The operating system is IBSYS. See section 4.4

4.1.3 Size. - The program is too large for an in-core program so the overlay feature of IBSYS is used to overcome this difficulty. The program is divided into three links, with each link being called once per data case.

4.1.4 Options. - The program allows the user the option of calculating the basic data or inputting the data directly. Also, optional input features are available. Following is a brief list of these options.

4.1.4.1 Optional $[\Phi A]^T$. - Three methods are allowed and they are as follows:

- (a) Input,
- (b) Calculated using theoretical modes,
- (c) Calculated using Fourier Series modes.

4.1.4.2 Optional $[\Phi^2]^T$. - Two methods are allowed and they are as follows:

- (a) Input,
- (b) Calculated using theoretical modes.

4.1.4.3 Optional Power Spectrum Input. - Three methods are allowed and they are as follows:

- (a) One PSD for each area in strip of shell,
- (b) One PSD for complete shell (homogenous pressure),
- (c) One PSD for each area of shell (angle-of-attack).

4.1.4.4 Optional Co-Spectrum Input. - Two methods are allowed and they are as follows:

- (a) Form one. See section 4.3.10.4
- (b) Form two. See section 4.3.10.4

4.1.5 Subroutines. - A brief description of the various subroutines and links used by DARC are given in the following paragraphs.

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4.1.5.1 TLU2, Linear Interpolation. - This routine calculates an $f(x)$ for x by a linear interpolation. No extrapolation will be made. If x is out of the range of the x values in the table an error indicator will be set. See section 4.6.3

4.1.5.2 DPRT, Prints Diagonal Matrix. - This routine prints a diagonal matrix as a row with proper identification. See section 4.6.4

4.1.5.3 DTRM, Calculates Results of Row Matrix Product. - This routine performs only those matrix operations necessary to determine the resulting elements of a row matrix times a rectangular matrix. See section 4.6.5.

4.1.5.4 EXPANC, Co-Spectra Expansion. - This routine expands the basic C matrix to be used in the form two expansion. See section 4.6.2

4.1.5.5 Main Program Link. - This link is used to call the necessary overlay links. See section 4.6.1

4.1.5.6 PART 1, Overlay Link 1. - This link is the first link called by the main program. The $[\Phi A]^T$, the $[\Phi^2]^T$, the integral of the Θ function, and the integral of the X function are calculated in this routine. See section 4.6.6

4.1.5.7 FHALF, Overlay Link 2. - This link is the second link called by the main program. It prints the matrices that are to be used (input or calculated) and reads in more tables for PSD and Co-Spectrum values. See section 4.6.7

4.1.5.8 SHALF, Overlay Link 3. - This link is the third link called by the main program. The program cycles and calculates: Normal Coordinate Deflection Power Spectra, Normal Coordinate Acceleration Power Spectra, Shell Radial Deflection Power Spectra, Shell Radial Acceleration Power Spectra, Mean-Square Normal Coordinate Deflection, Mean-Square Normal Coordinate Acceleration, Mean-Square Radial Deflection, and Mean-Square Radial Acceleration. This routine calls: TLU2, DPRT, DTRM, and EXPANC subroutines. See section 4.6.8

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4.2 General Program Flowcharts. - A condensed description of the entire flow of the program has been charted and is identified as Figure 4.2-1. It consists of 9 pages and contains many word descriptions rather than detailed equations.

GENERAL FLOW CHART

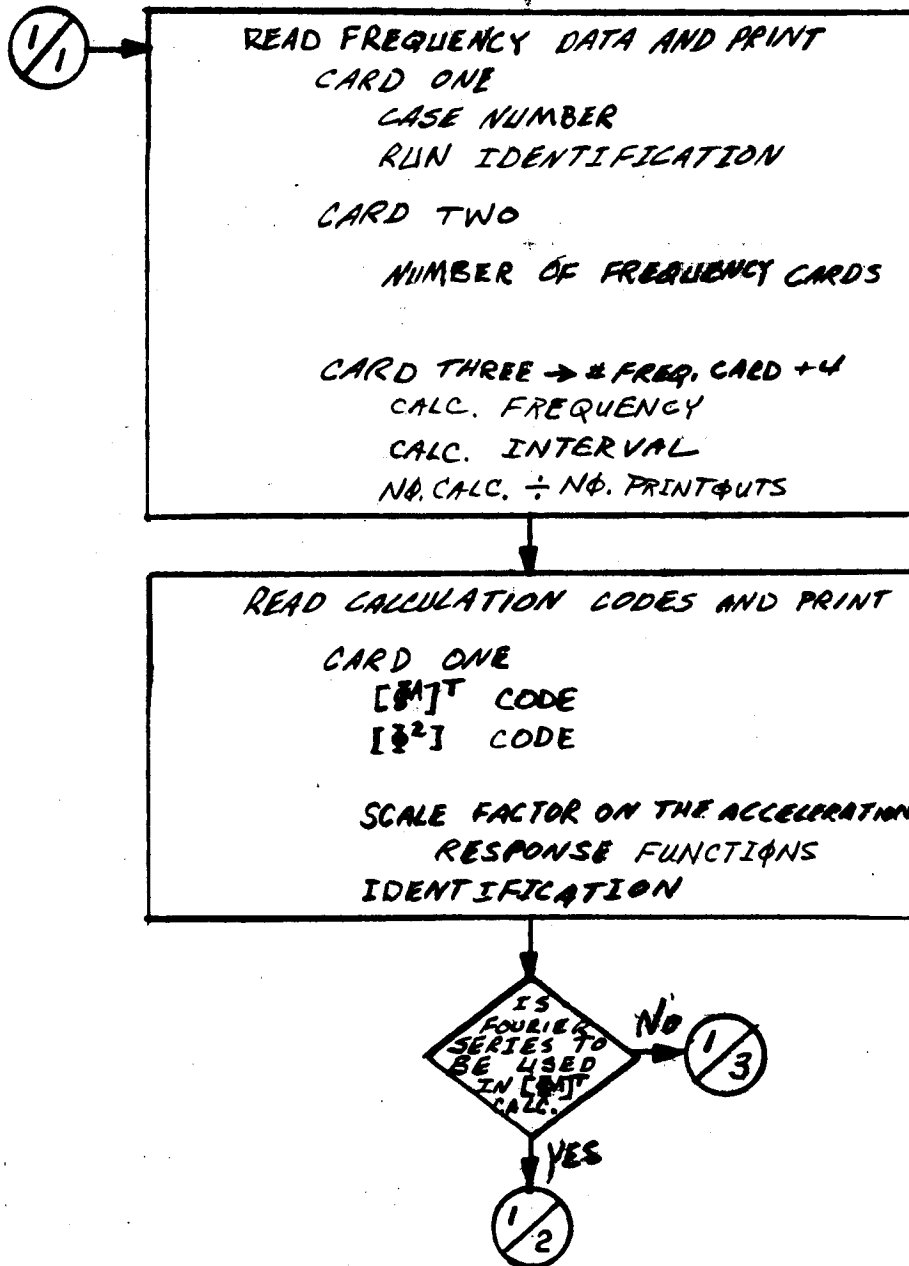


Figure 4.2-1

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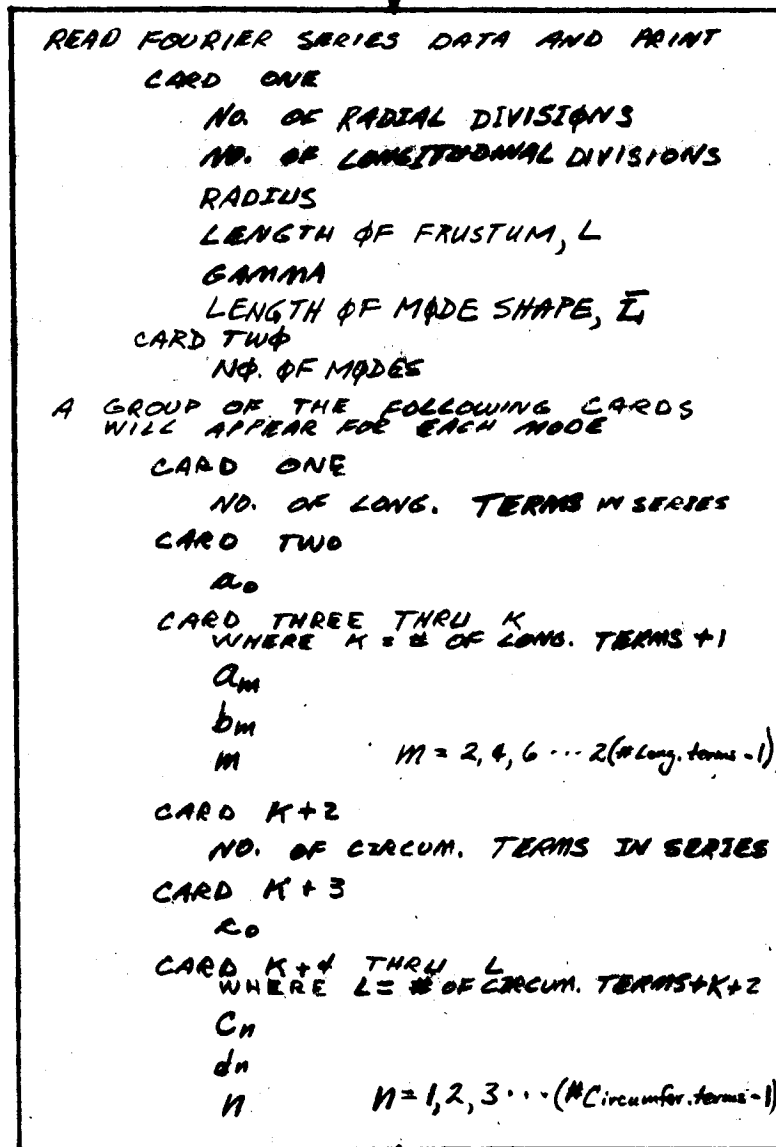
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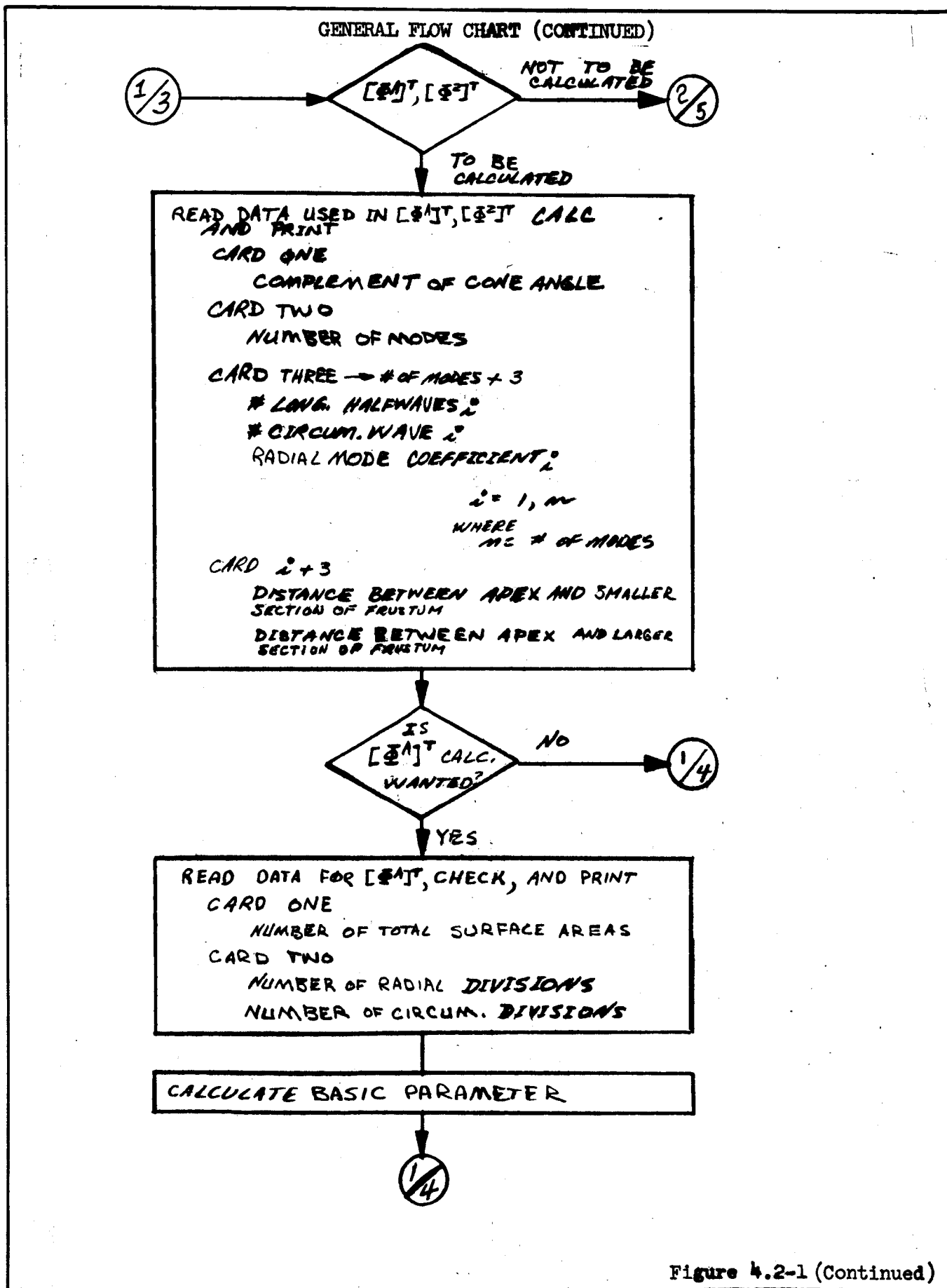
GENERAL FLOW CHART (CONTINUED)



CALCULATE $[Q_A]^T$ USING FOURIER SERIES



Figure 4.2-1 (Continued)



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GENERAL FLOW CHART (CONTINUED)

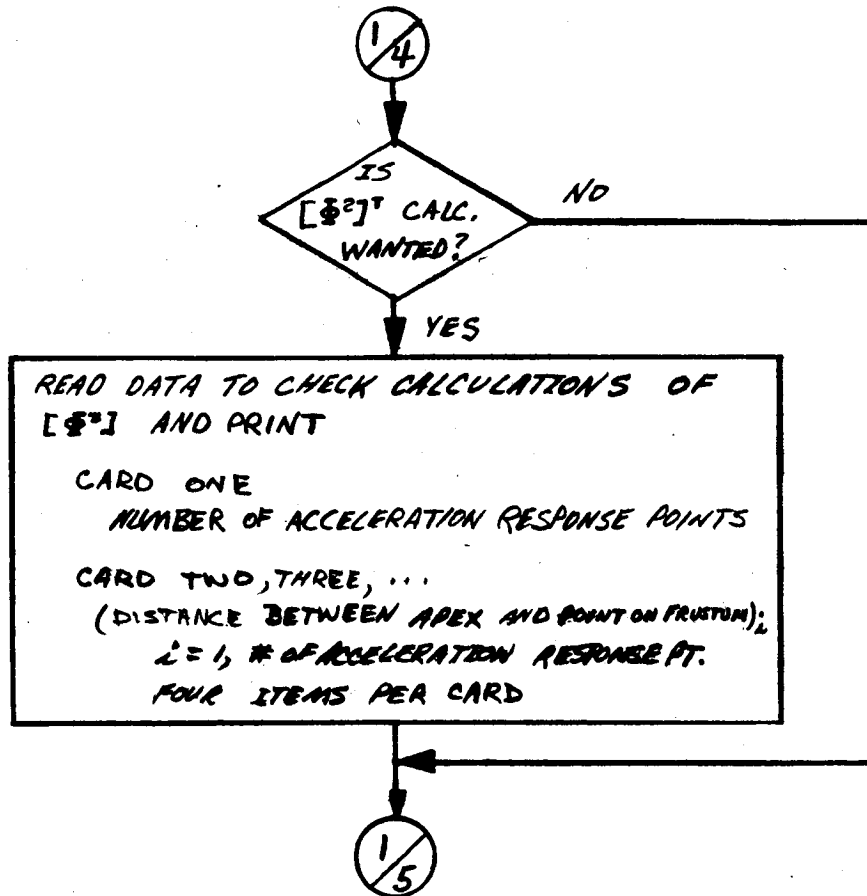


Figure 4.2-1 (Continued)

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GENERAL FLOW CHART (CONTINUED)

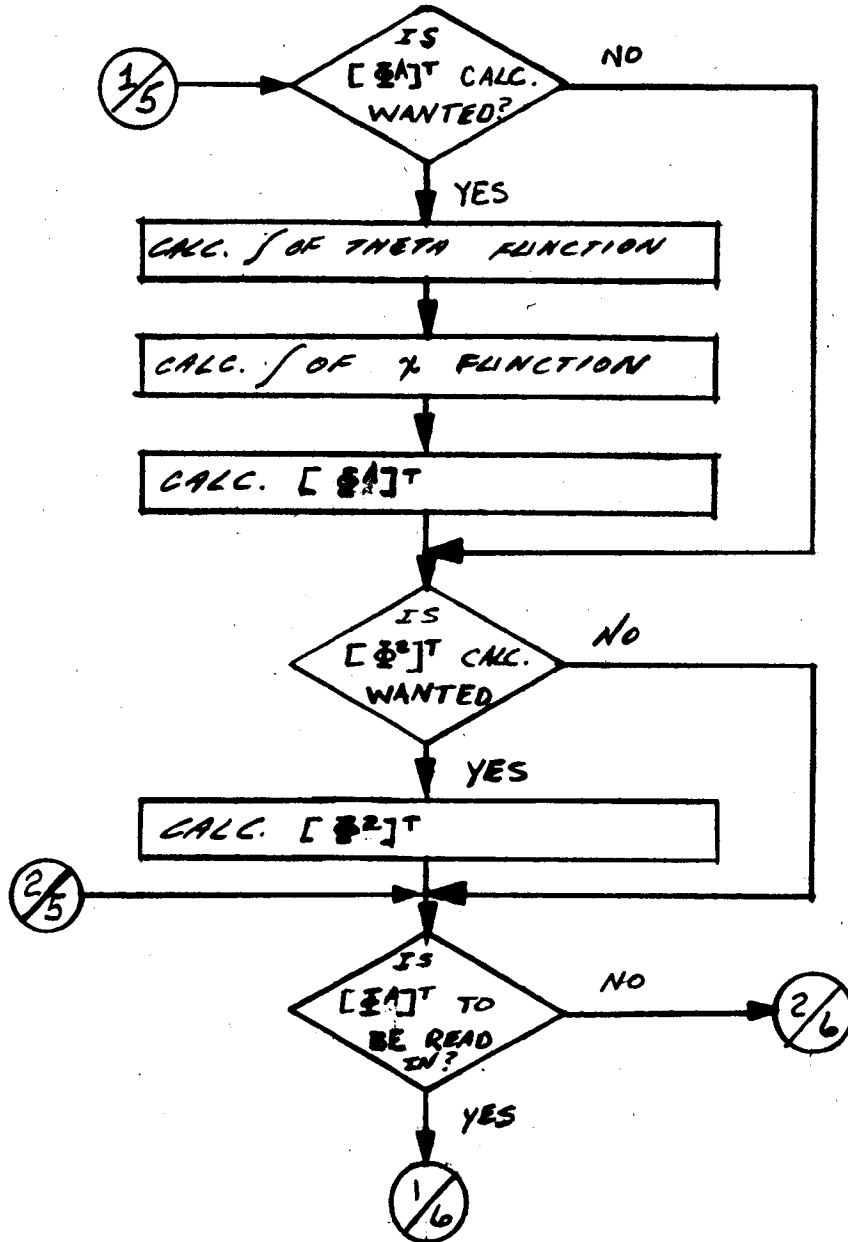


Figure 4.2-1 (Continued)

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GENERAL FLOW CHART (CONTINUED)

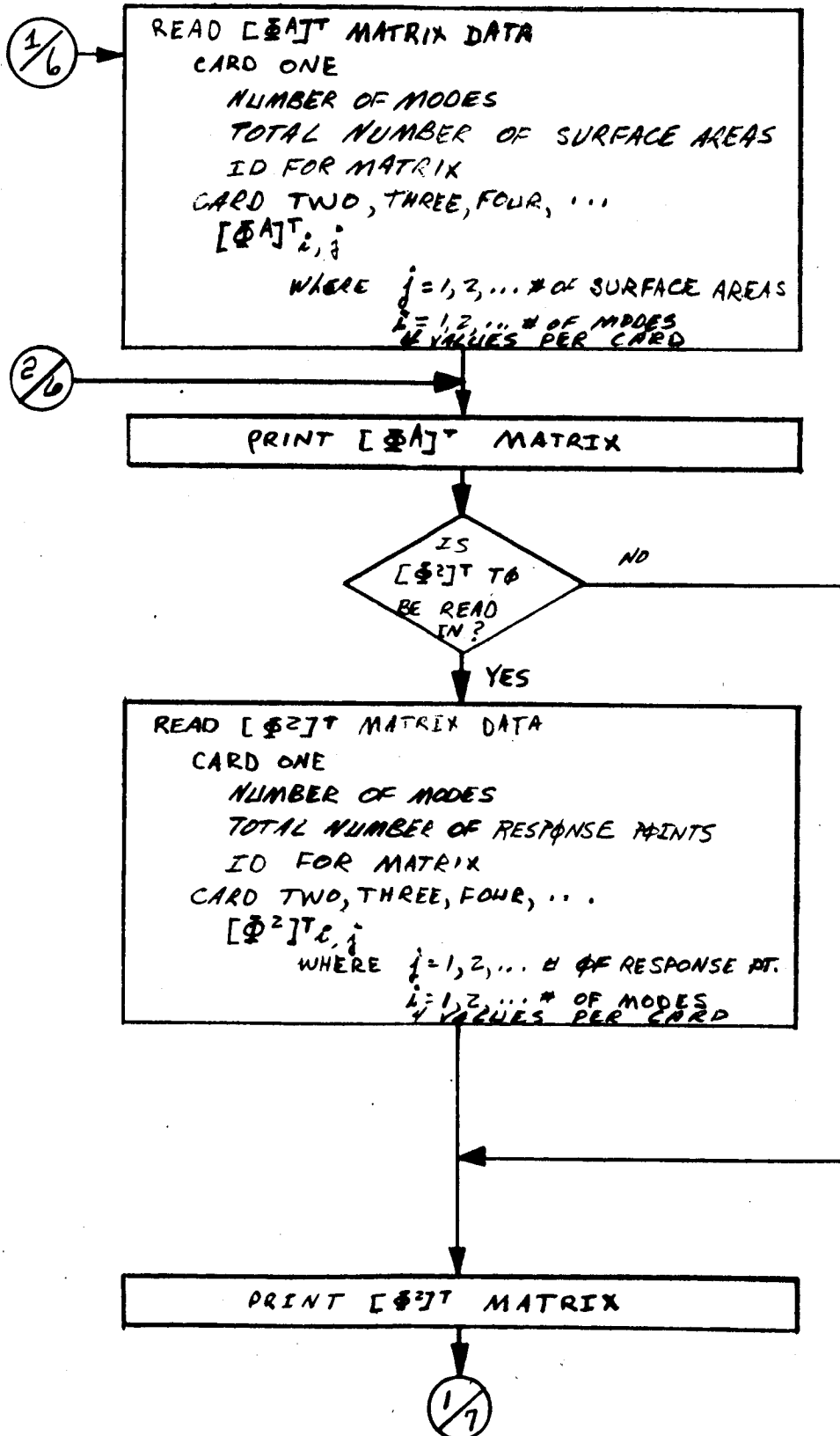


Figure 4.2-1 (Continued)

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GENERAL FLOW CHART (CONTINUED)

2/7

READ INPUT SPECTRUM DATA AND INPUT
CARD ONE

PSD INPUT CODE
CO-SPECTRUM EXPANSION CODE
EMPIRICAL CO-SPECTRA EQN CALC. CODE
SCALE FACTOR ON FREQUENCY
(FOR ALL SPECTRUM TABLES)
SCALE FACTOR ON PSD, ONLY

CARD TWO

NUMBER OF LONG. AREAS, KL
NUMBER OF LONG. AREAS OVER WHICH
EXISTS AN APPRECIABLE CORRELATION, KC
NUMBER OF CIRCUMFERENTIAL AREAS, KK

IS
CORRECTED
RESPONSE CALC.
REQUESTED

No

1/8

YES

READ EMPIRICAL CO-SPECTRA CALC. DATA
CARD ONE

DECAY RATE IN LONG. DIRECTION
DECAY RATE IN CIRC. DIRECTION
AVERAGE SIZE OF AREA (LONG.)
AVERAGE SIZE OF AREA (CIRC.)
CONVECTION VELOCITY

1/8

Figure 4.2-1(Continued)

GENERAL FLOW CHART (CONTINUED)

1/8

READ PSD DATA AND PRINT
 M # OF TABLES IN FOLLOWING PATTERN
 WHERE $M = 1$, IF PSD CODE = 1
 $M = \#$ OF LANG. AREAS, IF PSD CODE = 2
 $M = \#$ OF SURFACE AREAS, IF PSD CODE = 3
 CARD ONE
 NUMBER OF POINTS IN TABLE
 CARD TWO, THREE, ..., K
 (INDEPENDENT VARIABLE - FREQUENCY)
 WHERE $L = 1 \rightarrow \#$ PT. IN TABLE
 AT FOUR VALUES PER CARD
 CARD $K+1, K+2, \dots, M$
 (PSD) L, m
 WHERE $L = 1 \rightarrow \#$ PT. IN TABLE
 AT FOUR VALUES PER CARD

READ CO-SPECTRUM DATA AND PRINT
 M # OF TABLES IN FOLLOWING PATTERN
 WHERE $M = KK * KC$, FORM 1
 $M = KL * (KL-1)/2$, FORM 2
 (UPPER TRIANGLE OF MATRIX IS
 DISCUSSED IN ENGR. WRITE-UP)
 CARD ONE
 NUMBER OF POINTS IN TABLE
 CARD TWO, THREE, ..., K
 (INDEPENDENT VARIABLE - FREQUENCY) L, m
 WHERE $L = 1 \rightarrow \#$ PT. IN TABLE
 AT FOUR VALUES PER CARD
 CARD $K+1, K+2, \dots, M$
 (NORMALIZED CO-SPECTRUM DATA) L, m
 WHERE $L = 1 \rightarrow \#$ PT. IN TABLE
 AT FOUR VALUES PER CARD

1/9

Figure 4.2-1 (Continued)

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GENERAL FLOW CHART (CONCLUDED)

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READ NORMAL MODE INFORMATION AND PRINT
CARD ONE
NUMBER OF MODES
CARD TWO THRU K
WHERE $K = \text{NO. PARAMETERS} + 1$
(NORMAL MODE GENERALIZED MASS)
(NORMAL MODE DAMPING COEFFICIENTS)
(NORMAL MODE FREQUENCY)
WHERE $n=1, \dots, K$ OF PARAMETERS

INITIALIZE FREQUENCY CYCLE

EXPANSION OF PRESSURE CO-SPECTRUM MATRIX

MAKE MATRIX SYMMETRIC

CALCULATE Q DIAGONAL MATRIX

CALCULATE AND PRINT THE SPECIFIED
QUANTITIES

- 1) NORMAL COORDINATE DEFLECTION SPECTRA
- 2) NORMAL COORDINATE ACCELERATION SPECTRA
- 3) SHELL RADIAL DEFLECTION SPECTRA
- 4) SHELL RADIAL ACCELERATION SPECTRA

PERFORM INTEGRATION

CALCULATE NEW FREQUENCY

IS
NEW FREQ.
>
FINAL
FREQ.

PRINT
INTEGRATED
VALUES

1/1

Figure 4.2-1 (Concluded)

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4.3 Input Data Description. - Even though the DARC program deals with symmetrical pressures, it still requires a great deal of input data to adequately describe the problem. Herein is a detailed account of the formats required, the order of the data, and the numerical forms necessary.

4.3.1 Case Header Card. - Figure 4.3.1-1 shows the header card for each case. The case number is a fixed point integer and the case identification may be any alpha-numeric character desired.

4.3.2 Frequency Data. - Figure 4.3.1-1 also shows the format for inputting the calculation frequency data. First is a card specifying the number of cards to follow (input number must be fixed point). Second, then, is the set of frequency data cards. The maximum number of frequency data cards is 50. The values must be in floating point notation.

4.3.3 Matrix Control Card. - Figure 4.3.1-1 also shows the control card for determining $[\Phi^A]^T$ and $[\Phi^z]^T$. A value of one indicates direct input. Two indicates that the respective matrix is to be calculated using theoretical modes. Three indicates that $[\Phi^A]^T$ is to be calculated using Fourier Series modes. Three is not valid for $[\Phi^z]^T$. In addition a scale factor for multiplying the acceleration response function is included. This number must be floating point. An ID, or matrix comment, may also be input.

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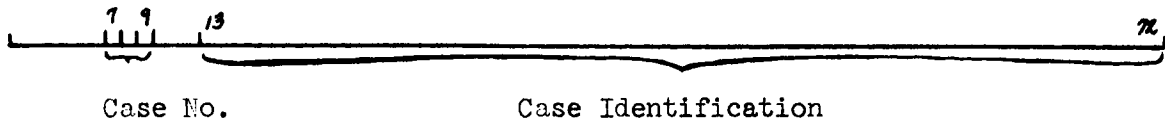
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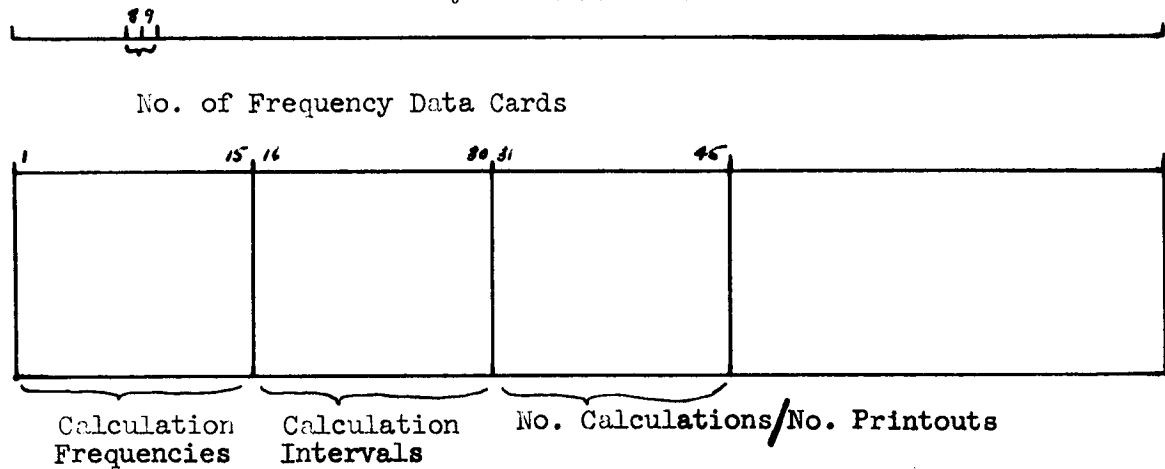
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MODEL _____

CASE HEADER CARD



FREQUENCY DATA CARDS



MATRIX CONTROL CARD

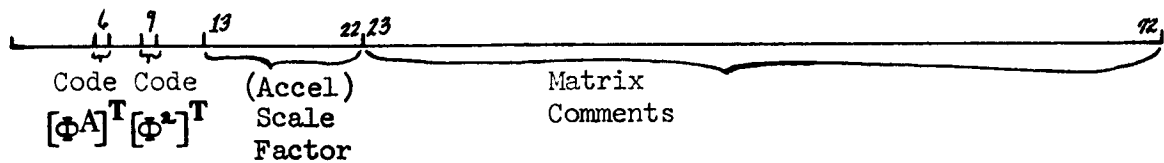


Figure 4.3.1-1

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4.3.4 Calculation of $[\Phi^A]^T$ By Fourier Series. - If the code specified is a three, then the set of data described in Figure 4.3.4-1 must be included. Otherwise this portion must be omitted.

4.3.4.1 Fourier Series Basic Data. - In Figure 4.3.4-1 the first card contains two fixed point integers denoting divisions of the frustum or cylinder and four floating point parameters. Card two contains the number of modes (maximum of 25). Following this, then, is a dual set of coefficients for each mode. They are described in the following paragraphs. The longitudinal coefficients are always first.

4.3.4.2 Longitudinal Coefficients. - In Figure 4.3.4-1, under Longitudinal Coefficients, the first card contains a fixed point integer denoting the number of coefficients for the longitudinal series. The maximum number of coefficients is 31. Following this is a set of cards containing the coefficients with the first card containing R_0 . The coefficients relating to the Fourier Series are defined in Equation 21, Section 3.3.2.

4.3.4.3 Circumferential Coefficients. - In Figure 4.3.4-1, under Circumferential Coefficients, the first card contains a fixed point integer denoting the number of coefficients for the circumferential series. The maximum here, too, is 31. Following this is a set of cards containing the coefficients with the first card containing R_0 . The coefficients relating to the Fourier Series are defined in Equation 21, Section 3.3.2.

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FOURIER SERIES BASIC DATA

<u>7 9 10 12</u> <u>16</u> No. of Longitudinal Divisions No. of Circumferential Divisions	<u>30 31</u> Average Radius of Frustum or Cylinder	<u>45 46</u> Length of Frustum or Cylinder	<u>60 61</u> Origin of Mode Shape less Origin of Frustum or Cylinder (Must be positive)	<u>75</u> Length of Mode Shapes
--	---	---	---	---------------------------------------

<u>7 9</u> No. of Modes	
----------------------------	--

LONGITUDINAL COEFFICIENTS

<u>8 9</u> No. of Coefficients	
-----------------------------------	--

<u>1</u> 	<u>15 16</u> 	<u>30 31</u> 	<u>45</u>
a	b	m	

CIRCUMFERENTIAL COEFFICIENTS

<u>8 9</u> No. of Coefficients	
-----------------------------------	--

<u>1</u> 	<u>15 16</u> 	<u>30 31</u> 	<u>45</u>
c	d	n	

(Repeat both Longitudinal and Circumferential Coefficients for each mode).

Figure 4.3.4-1

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4.3.5 Calculation of Either $[\Phi^A]^T$ Or $[\Phi^2]^T$ By The Theoretical Modes. - If the code specified for either $[\Phi^A]^T$ or $[\Phi^2]^T$ is a two, then the set of data described in Figure 4.3.5-1 must be included. Otherwise this portion should be omitted.

4.3.5.1 Theoretical Modes Basic Parameter Card. - In Figure 4.3.5-1, under Theoretical Modes Basic Parameter Card, a single card is described which denotes an angular constant.

4.3.5.2 Theoretical Modes Data. - In Figure 4.3.5-1, under Theoretical Modes Data, the first card contains a fixed point number denoting the number of data cards to follow. Following it, then, is a set of data cards in floating point notation. The maximum number of theoretical modes is 25.

4.3.5.3 Distance Parameter Card. - In Figure 4.3.5-1, under Distance Parameter Card, a single card is described which denotes two floating point constants. These are used by the Theoretical Modes Option.

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THEORETICAL MODES BASIC PARAMETER CARD

1 15

$\bar{\alpha}$, Complement of the Cone Angle

THEORETICAL MODES DATA

7 9

No. of Theoretical Mode Data Cards

1	10, 11	20, 21	35

Radial Mode Coefficients (A)
No. of Longitudinal Half-waves (m)
No. of Circumferential Full-waves (n)

DISTANCE PARAMETER CARD

1 15, 16 30

Distance Between Apex of Cone & Large Diameter
of Frustum
Distance Between Apex of Cone & Small Diameter of Frustum

Figure 4.3.5-1

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4.3.6 Calculation of $[\Phi^A]^T$ Using Theoretical Modes. - When the code specified is a two, then the two data cards described in Figure 4.3.6-1 must be included. Otherwise this portion should be omitted.

4.3.6.1 Area Parameter Cards. - In Figure 4.3.6-1, under Area Parameter Cards, the first card contains a fixed point parameter. The maximum value is 112. The second card contains two parameters in fixed point notation. The product of the two numbers must equal the number of the first card. In addition to that, the parameter which defines the number of longitudinal areas is limited to a maximum of seven.

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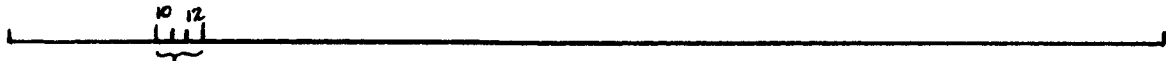
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AREA PARAMETER CARDS



No. of Surface Areas



No. of Longitudinal Areas
No. of Circumferential Areas

Figure 4.3.6-1

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4.3.7 Calculation of $[\Phi^*]^T$ Using Theoretical Modes. - When the code specified is a two, then the set of data cards described in Figure 4.3.7-1 must be included. Otherwise this portion should be omitted.

4.3.7.1 Acceleration Response Points Desired. - In Figure 4.3.7-1, under Acceleration Response Points Desired, the first card contains a fixed point parameter denoting the number of points specified. Following it, then, is a set of cards containing four floating point values per card. This data represents the distances measured from the apex. The maximum number of response points is 20.

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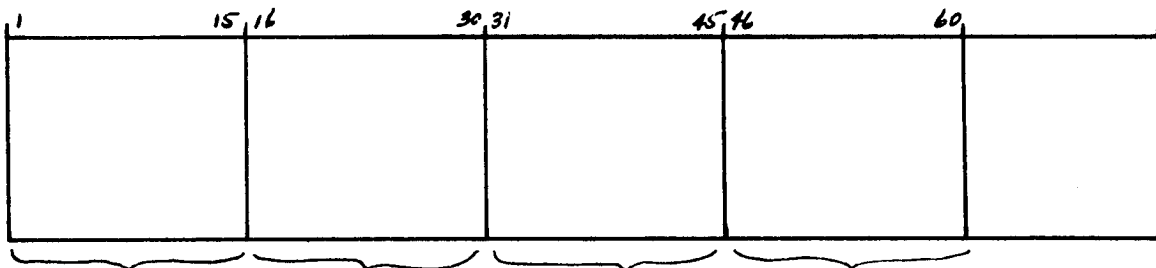
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ACCELERATION RESPONSE POINTS DESIRED



No. of Acceleration Response Points



Each value is distance measured from the apex.

Figure 4.3.7-1

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4.3.8 Input $[\Phi A]^T$. - When the code specified is a one, then the set of data cards described in Figure 4.3.8-1 must be included. Otherwise this portion should be omitted.

4.3.8.1 $[\Phi A]^T$ Data Cards. - In Figure 4.3.8-1, under $[\Phi A]^T$ Data Cards, the first card contains a pair of fixed point numbers denoting the order of the matrix to be input. Following it, then, is a set of cards containing four floating point values per card. The matrix is input rowwise with each row beginning on a new card. The maximum order is (25,112). The elements of each row equal the number of surface areas. Each row pertains to a different mode.

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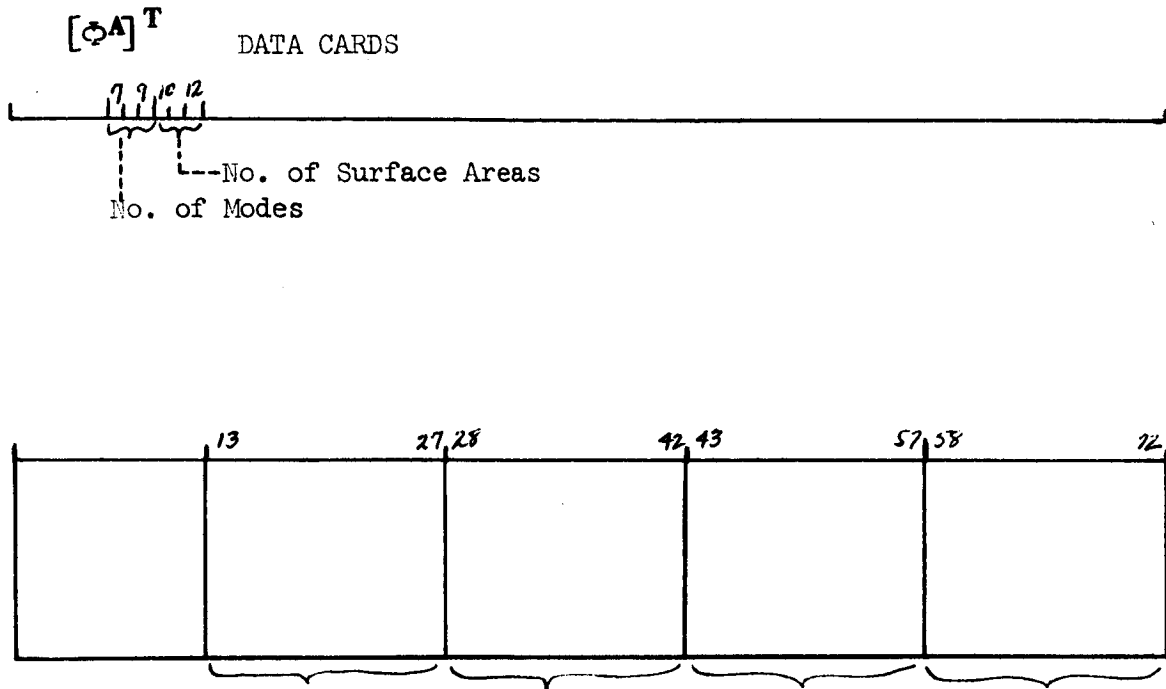
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Matrix values with four elements per card, input rowwise,
and each row begins on a new card.

Figure 4.3.8-1

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4.3.9 Input $[\Phi^2]^T$. - When the code specified is a one, then the set of data cards described in Figure 4.3.9-1 must be included. Otherwise this portion should be omitted.

4.3.9.1 $[\Phi^2]^T$ Data Cards. - In Figure 4.3.9-1, under $[\Phi^2]^T$ Data Cards, the first card contains a pair of fixed point numbers denoting the order of the matrix to be input. Following it, then, is a set of cards containing four floating point values per card. The matrix is input rowwise with each row beginning on a new card. The maximum order is (25,20). The number of elements of each row equals the number of response points desired. Each row pertains to a different mode.

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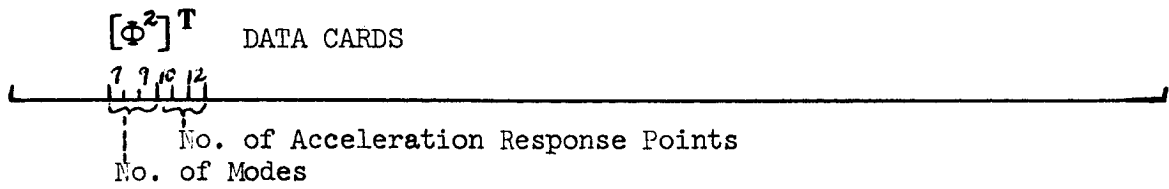
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	13	27 28	42 43	57 58	72

Matrix values with four elements per card, input rowwise, and each row begins on a new card.

Figure 4.3.9-1

4.3.10 Spectra Input Data. - The power spectral densities (PSDs) and the normalized co-spectra are represented in the DARC program in the form of tables. Linear interpolation is used to obtain the spectrum at a desired frequency. There exists several options of input. This, and other necessary information, will be described in the following paragraphs.

4.3.10.1 Table Control Cards. - Figure 4.3.10.1-1 shows the two control cards required to define the option of input and scalars used.

4.3.10.1.1 PSD Input Code. - In Figure 4.3.10.1-1 the first card contains the PSD Input Code. The code definitions are:

(a) A code of one specifies that a PSD is required for each area in one longitudinal strip. This information is then used for all other strips.

(b) A code of two specifies that one PSD is sufficient for the complete shell (Homogeneous Pressure Distribution).

(c) A code of three specifies that a PSD is required for each area of the shell (Angle of Attack).

4.3.10.1.2 Normalized Co-Spectra Expansion Code. - In Figure 4.3.10.1-1 the first card contains the Normalized Co-Spectra Expansion Code. The code definitions are:

(a) A code of zero specifies that form one is used. See section 4.3.10.4

(b) A code of one specifies that form two is used. See section 4.3.10.4

4.3.10.1.3 Empirical Co-Spectra Calculate Code. - In Figure 4.3.10.1-1 the first card also contains a code specifying the use of empirical co-spectrum rather than using a direct table-lookup. If the code specifies zero, this means no calculation. A non-zero value activates the option.

4.3.10.1.4 Table Scale Factors. - In Figure 4.3.10.1-1 the remaining two floating point parameters of the first card are scale factors. The first applies to all frequency values in all the tables. The latter applies to the PSD magnitudes only.

4.3.10.1.5 Table Input Control Card. - In Figure 4.3.10.1-1 the second card contains three fixed point numbers which specify the number of tables to be input. The parameters, as previously described, are KL, KC, and KK in that order.

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TABLE CONTROL CARDS

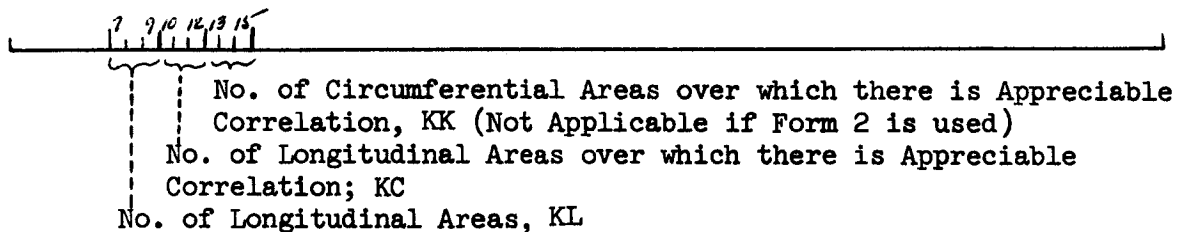
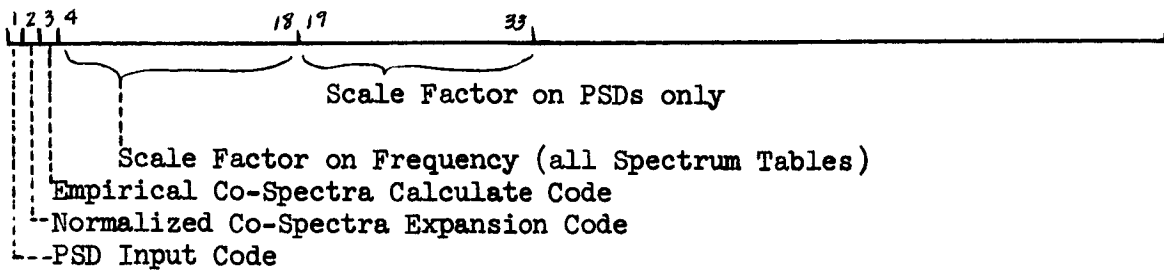


Figure 4.3.10.1-1

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4.3.10.2 Empirical Co-Spectra Calculation Parameters. - If the code specified is non-zero, then the data card described in Figure 4.3.10.2-1 must be included. Otherwise this card should be omitted.

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MODEL _____

EMPIRICAL CO-SPECTRA CALCULATION PARAMETERS

<u>1</u>	<u>15, 16</u>	<u>30, 31</u>	<u>45, 46</u>	<u>60, 61</u>	<u>75</u>
Longitudinal Decay Rate (1/in.)	Circumferential Decay Rate (1/in.)	Average Size of long. area (in.)	Average Size of circ. area (in.)	Convection Velocity (ft./sec.)	

Figure 4.3.10.2-1

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4.3.10.3 PSD Tables. - In Figure 4.3.10.3-1 the first card contains a fixed point number denoting the number of points for that table. Following it, then, are two sets of data cards in floating point notation. The first set, with four values per card, contains the independent variable, frequency. The second set, also with four values per card, contains the dependent variable, PSD. The maximum number of tables is 112. The number of PSD tables to be input is based on the PSD Input Code. The following rules apply:

(a) If the code is one, then the number of tables input equals the number of longitudinal areas.

(b) If the code is two, then only one table is input.

(c) If the code is three, then the number of tables input equals the number of surface areas.

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PSD TABLES

89
111

No. of Points in Table

1	15, 16	30, 31	45, 46	60

Frequency Values (four per card)

1	15, 16	30, 31	45, 46	60

PSDs (four per card)

(Repeat process for each PSD table input)

Figure 4.3.10.3-1

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4.3.10.4 Normalized Co-Spectra Tables. - In Figure 4.3.10.4-1 the first card contains a fixed point number. If this number is less than or equal to 20, then the number represents the number of points for that table. Following it, then, are two sets of data cards in floating point notation. The first set, with four values per card, contain the independent variable, frequency. The second set, also with four values per card, contain the dependent variable, Co-Spectrum. The maximum number of points is 20. The maximum number of tables is 21. The number of Co-Spectra tables to be input is based on the Co-Spectra Expansion Code. The following rules apply:

(a) If the code is zero, then input the tables for the longitudinal areas by column, including the (1,1) element, up to KC, which is the number of longitudinal areas over which there is appreciable correlation. Repeat for KK columns where KK is the number of circumferential areas over which there is appreciable correlation. Thus, there are $KK \times KC$ tables. See section 3.2.1.1 for physical explanation.

(b) If the code is one, then input the tables for the upper right triangle of the matrix by row (excluding the diagonal). Thus, the number of tables equals $\frac{KL(KL-1)}{2}$, where KL is the number of longitudinal areas. See section

3.2.1.2 for physical explanation.

4.3.10.4.1 Element-By-Element Calculation Code. - In paragraph 4.3.10.4 a description was given for the input of a co-spectrum table. If the number of points specified exceeds 20, then no table should be input. This feature along with the Empirical Co-Spectra Calculate Code activated, results in calculating the normalized co-spectrum for this element. If the Empirical Co-Spectra Calculate Code is activated, then the (1,1) element is always calculated. See section 3.2.2.1 and subsections for physical explanation.

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CO-SPECTRA TABLES

89

No. of Points in Table

1	15 16	30 31	45 46	60

Frequency Values (four per card)

1	15 16	30 31	45 46	60

Normalized Co-Spectrum (four per card)

(Repeat process for each Normalized Co-Spectrum table input)

Figure 4.3.10.4-1

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4.3.11 Normal Mode Parameters. - In Figure 4.3.11-1 the first card contains a fixed point number denoting the number of data cards to follow. Following it, then, is the set of floating point data. The maximum number of data cards is 25.

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NORMAL MODE PARAMETERS

$\overbrace{\hspace{10em}}^{89}$
 $\underbrace{\hspace{10em}}_{15}$
 No. of Modes

1	15, 16	30, 31	45
$\underbrace{\hspace{10em}}$ Normal Mode Generalized Mass	$\underbrace{\hspace{10em}}$ Normal Mode Damping Coefficient	$\underbrace{\hspace{10em}}$ Normal Mode Frequency	

Figure 4.3.11-1

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4.4 Operational Procedures. - This section contains two basic operational requirements. They are operating instructions and deck configuration.

4.4.1 Operating Instructions. - The basic computer setup is described on the following page. It should be noted that time estimates are also included.

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OPERATING INSTRUCTIONS

DECK NO.	TITLE				
T650	Conical Shell Deflection and Acceleration Response Program				
PROGRAMMER			ASSIGNED TO		
A. Brunkow/ S. Linn, Dept. 73, MAC			S. Linn		
REQUESTED BY				DEPT.	C.R. NO.
Dept. 236, MAC (Contract NAS 1-3179)					
REPLACES DECK	NEW DECK	MODIFICATION OF DECK	☐ SOS	☐ FORTRAN II	☒ IB.SYS
	☒				☐ OTHER
RESERVED TAPES USED					
None					
UTILITY TAPES USED					
None					
OUTPUT					
TAPES	Standard Fortran Printout Tape				
CARDS	PLOTting		INPUT TO OTHER DECK		OTHER
☐	☐		☐		☐
DEFINE CARD FORMAT BELOW					
1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20 21 22 23 24 25 26 27 28 29 30 31 32 33 34 35 36 37 38 39 40 41 42 43 44 45 46 47 48 49 50 51 52 53 54 55 56 57 58 59 60 61 62 63 64 65 66 67 68 69 70 71 72 73					
EST. NO. OF INPUT CARDS PER CASE					
<u>50/1500 (Depends on options requested)</u>					
EST. RUNNING TIME PER CASE					
<u>.02/15 hr (7094)</u>					
EST. PRINTING TIME PER CASE					
<u>.05/.40 hr (1401)</u>					
CAN THIS JOB BE INTERRUPTED					
☒ YES ☐ NO					
CAN MORE THAN ONE CASE BE RUN AT A TIME					
☒ YES ☐ NO					
SPECIAL INSTRUCTION:					
None					

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4.4.2 Deck Configuration. - The basic deck setup is shown on the following pages. The control cards for IBSYS at the M.A.C. installation consists of two cards. They are shown as the first two cards of the program listing in section 4.8

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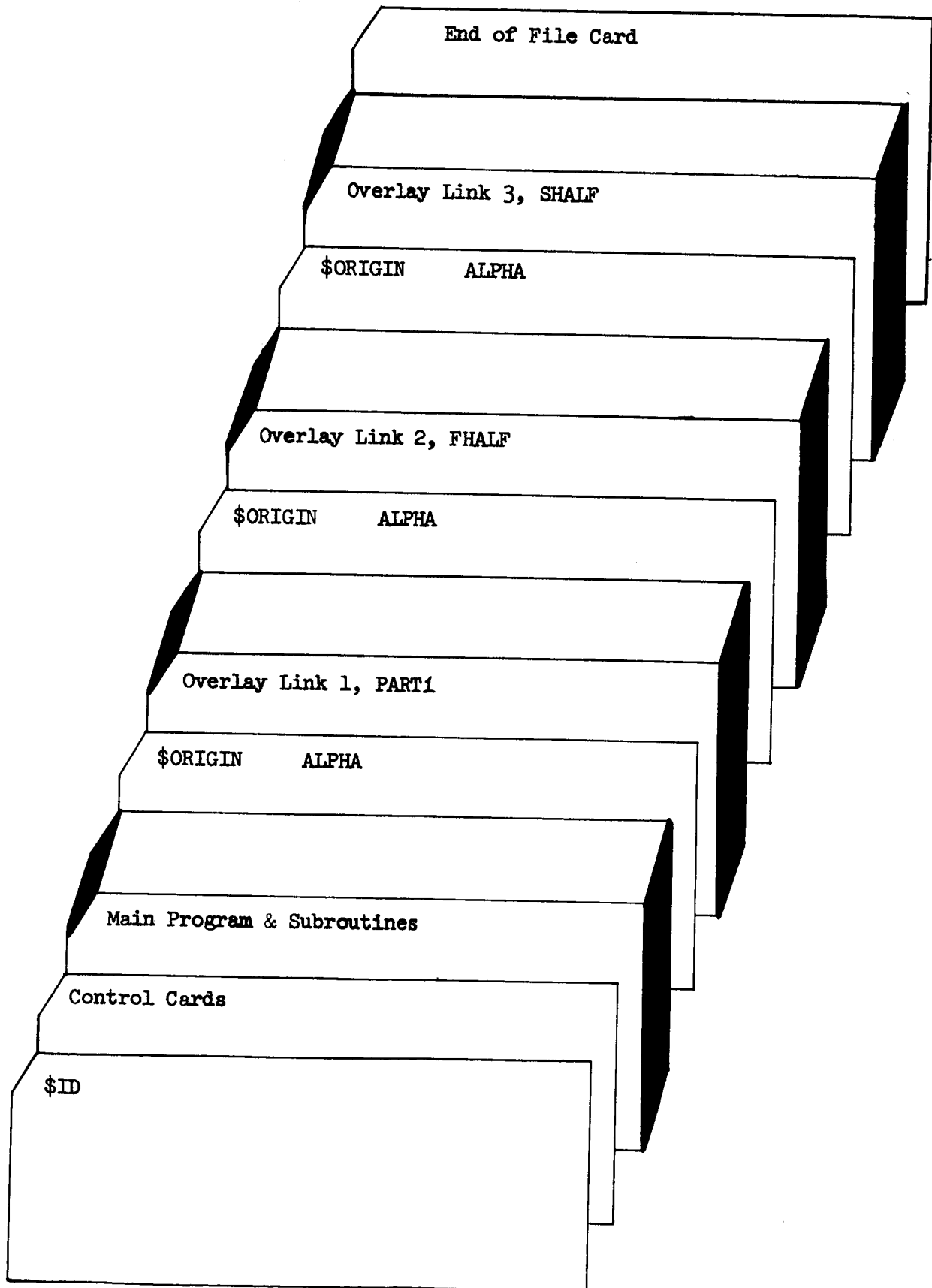
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DECK ORGANIZATION

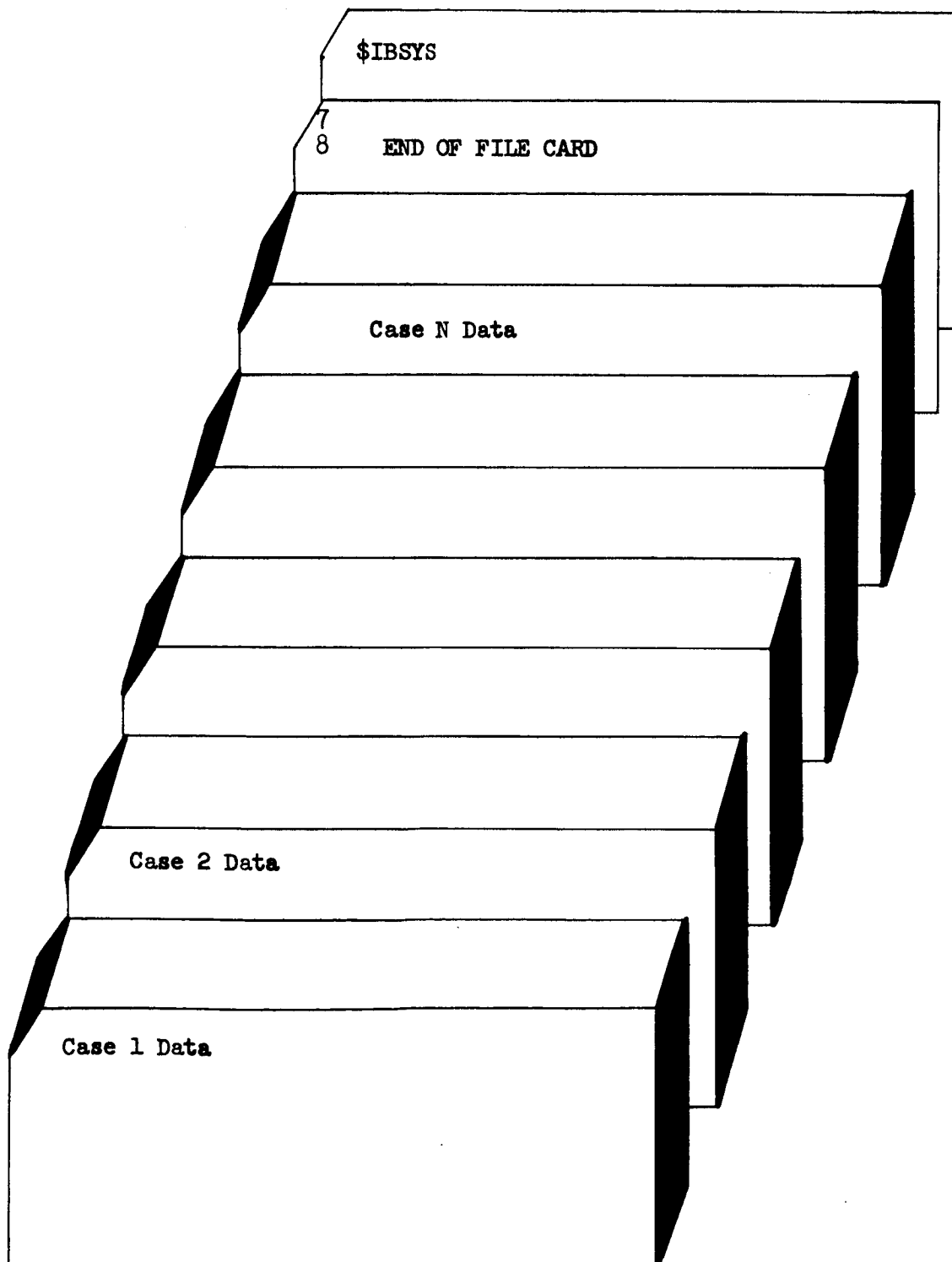
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DECK ORGANIZATION (Concluded)

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4.5 Error Analysis and Messages - There exists in the program various error checks before calculations actually begin. When an error is detected by the program a comment is written on the output tape and appropriate action is taken. Following is a list of error comments and their meanings.

4.5.1 Program Cannot Continue, Gamma = X.XXXXXXX - This is the result of a negative gamma to be used in the Fourier Series Calculation. The program terminates.

4.5.2 Response Calculation Involving PHI2T & PHIAT Depends Upon Assumption of Symmetric Pressure Power Spectra Distribution. LCOD = Indicates Unsymmetric Distribution - This is self-explanatory. The comment is merely a warning.

4.5.3 Error, PHI2T & PHIAT Incompatible, XXX YYY - The number of rows in PHIAT (XXX) must be equal to the number of rows in PHI2T(YYY). The error, XXX not equal to YYY, terminates the program.

4.5.4 Error, PHIAT & Normal Modes Unequal, XXX YYY - The number of normal modes specified must be the same as the number of rows of PHI2T. The error terminates the program.

4.5.5. Error, TLU ERR In XXX Tables - In generating the spectrum matrices at a particular frequency, a total of XXX errors occurred in the linear interpolation routine (TLU2). This includes the PSDs and the co-correlation tables. The error terminates the program.

4.5.6 General Errors. - The system may output error comments. These usually result from data setup errors. These errors cause the system to terminate the program.

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4.6 Detailed Flowcharts. - Extensive program flow charts are given in this section. The blocks contain mostly Fortran terminology to closely identify the flowcharts with the program. Also included for this purpose were: statement numbers, DO loops, and sub-titles.

4.6.1 Main Program. - Figure 4.6.1-1 shows the linkage of the various sub-programs as they are executed by the main program.

MAIN PROGRAM

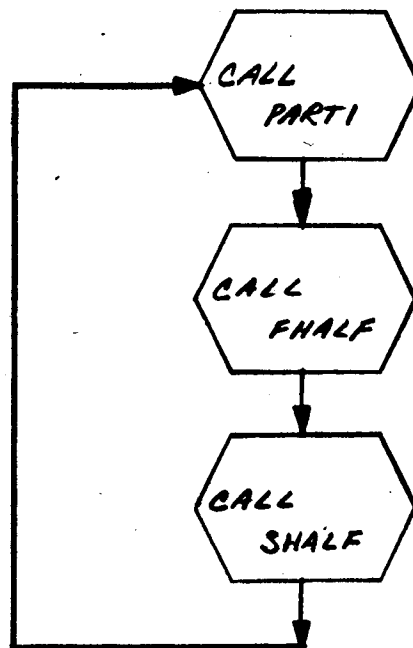


Figure 4.6.1-1

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4.6.2 Subroutine EXPANC. - Figure 4.6.2-1 shows the calling sequence and flow of the EXPANC subroutine. Note that this is for expansion form two.

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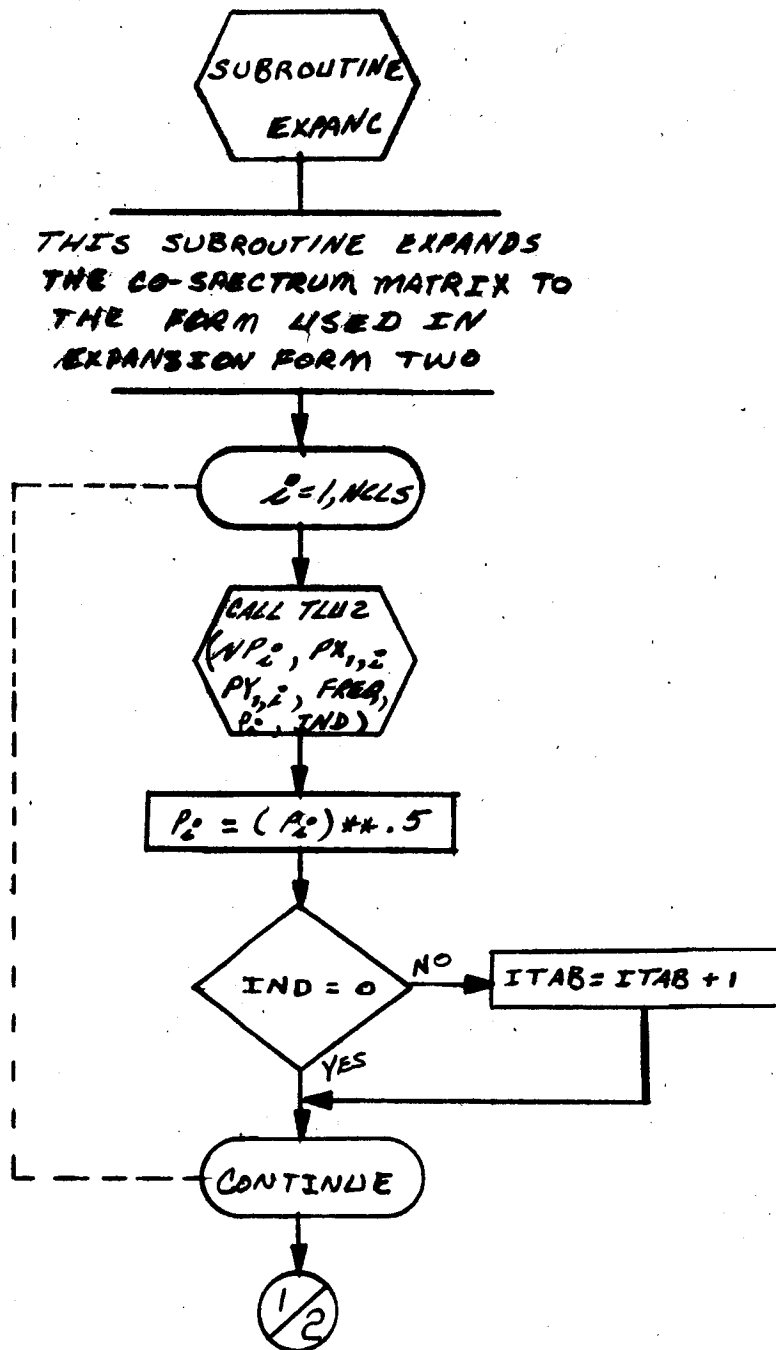


Figure 4.6.2-1

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Subroutine EXPANC (Concluded)

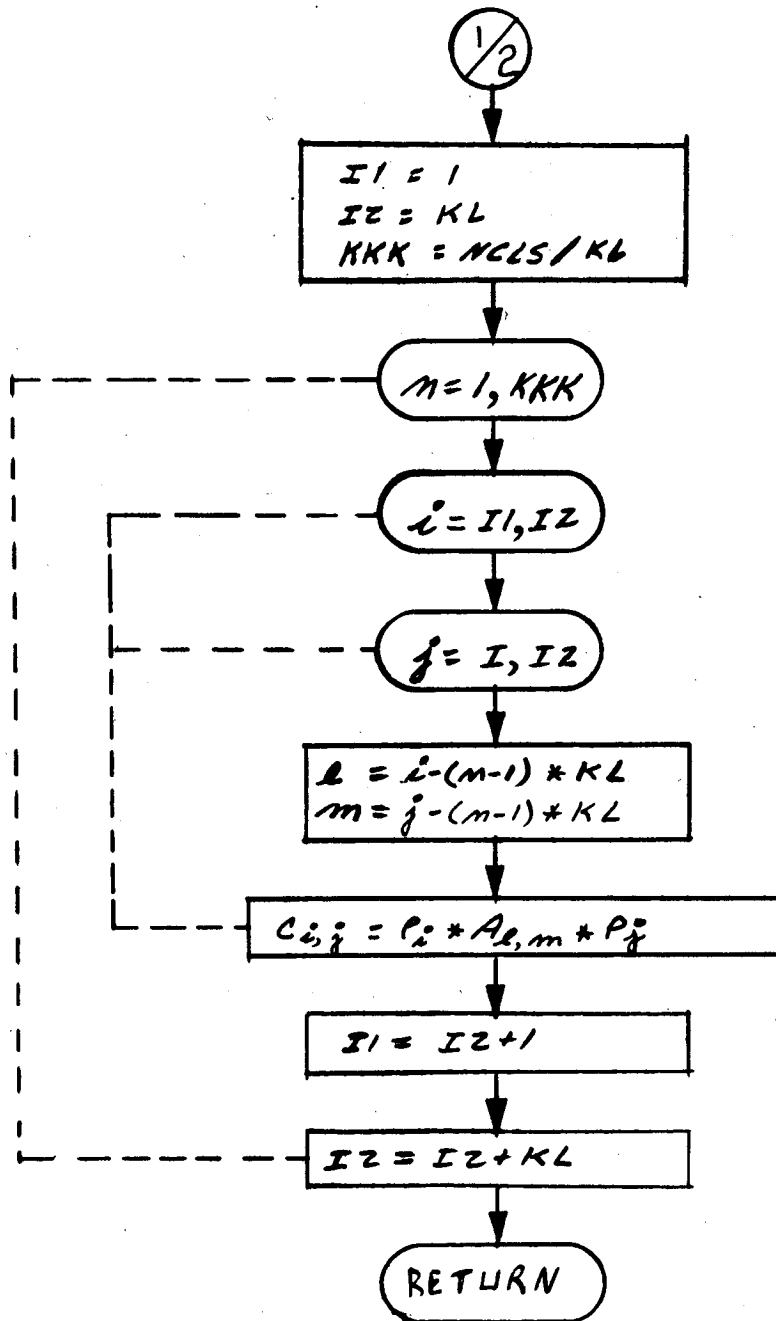


Figure 4.6.2-1(Concluded)

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4.6.3 Subroutine TLU2. - Figure 4.6.3-1 shows the calling sequence and flow of the table lookup routine, TLU2.

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SUBROUTINE
TLU2 (N,XV,YV,X,Y,IND)

LINEAR INTERPOLATION ROUTINE

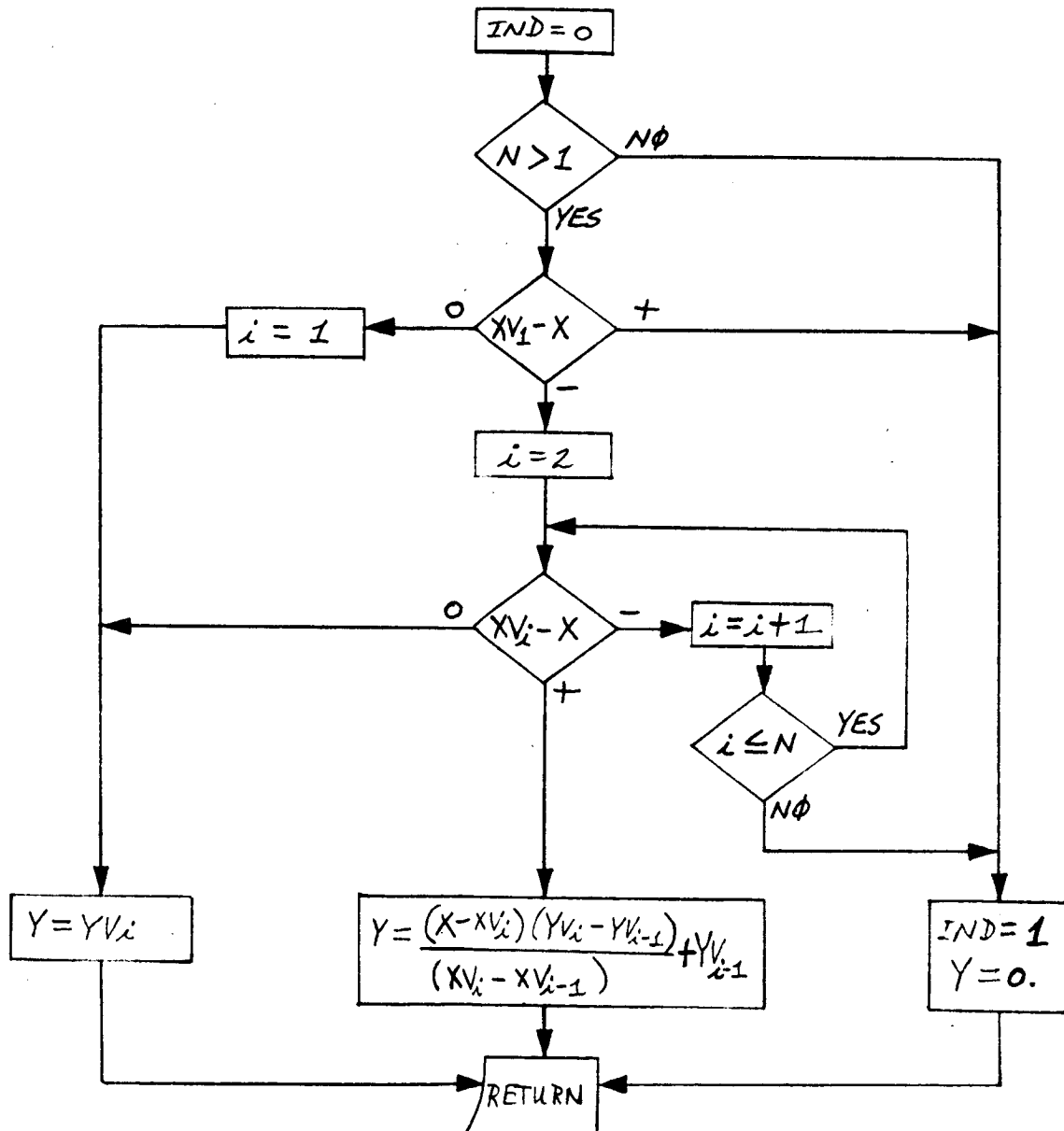


Figure 4.6.3-1

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4.6.4 Subroutine DPRT. - Figure 4.6.4-1 shows the calling sequence and flow of the diagonal matrix print subroutine, DPRT.

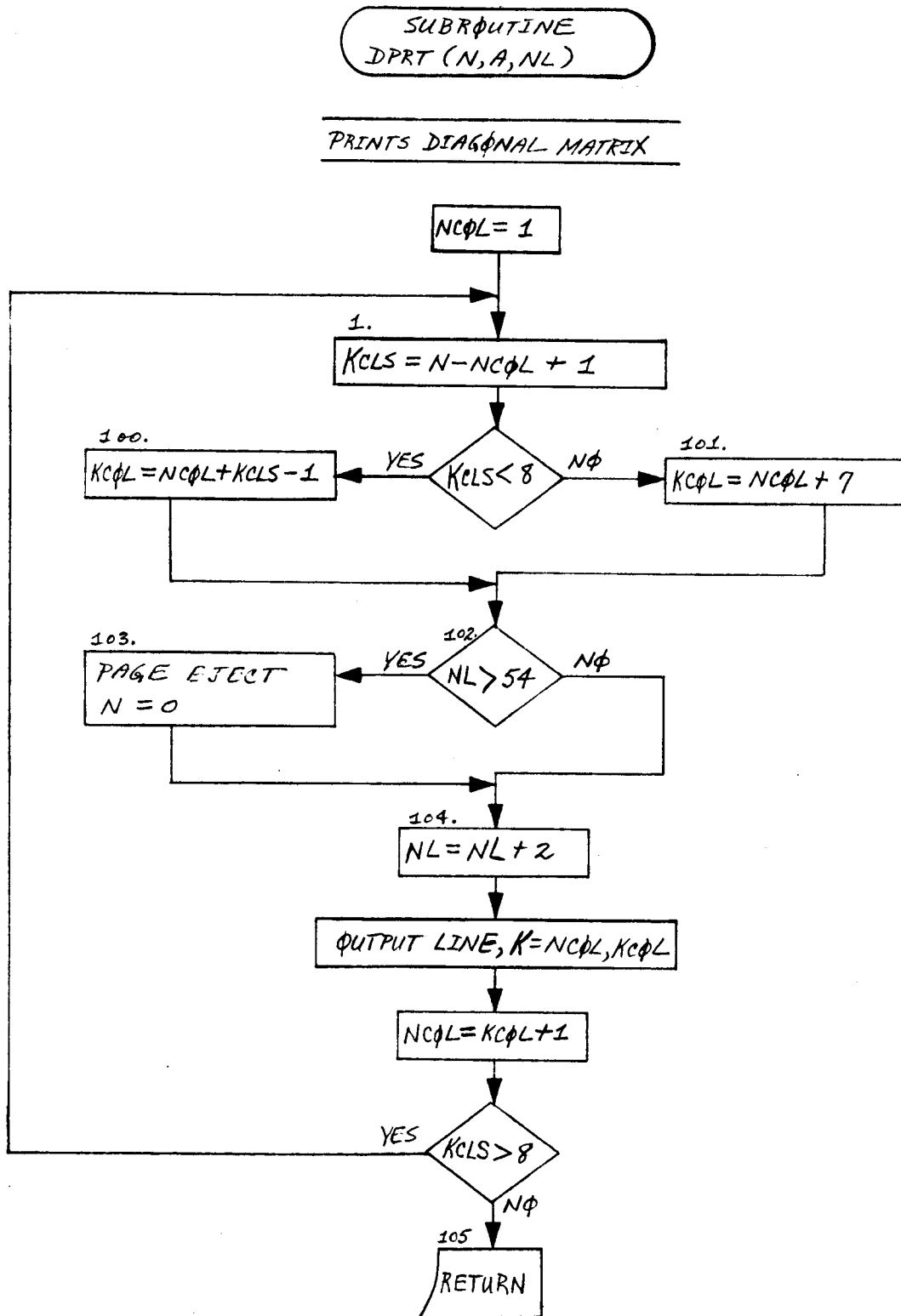


Figure 4.6.4-1

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4.6.5 Subroutine DTRM. - Figure 4.6.5-1 shows the calling sequence and flow of the diagonal matrix product subroutine, DTRM. It calculates product row A * [P].

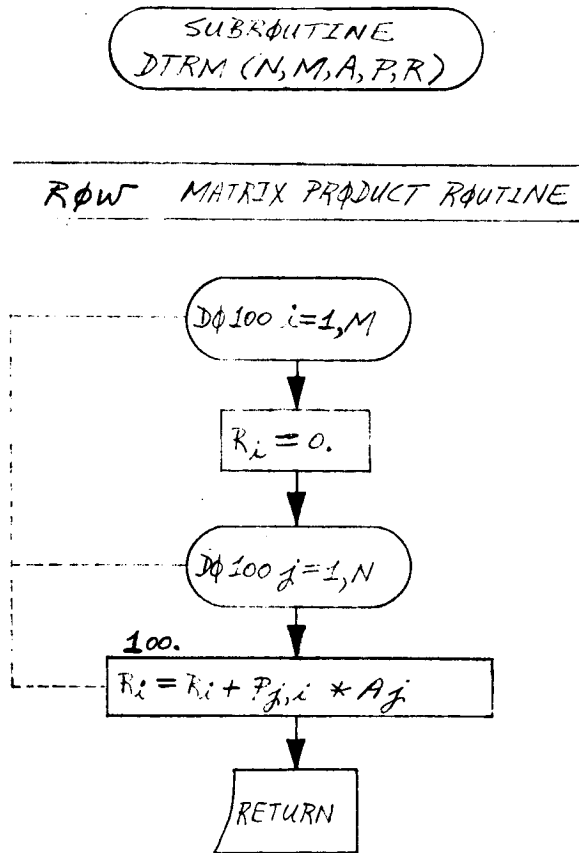


Figure 4.6.5-1

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4.6.6 Link One, PART 1. - Figure 4.6.6-1 shows the initial input portion for each case. It consists of 14 pages. After this link is executed and completed the program returns control to the MAIN program.

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Subroutine PART1

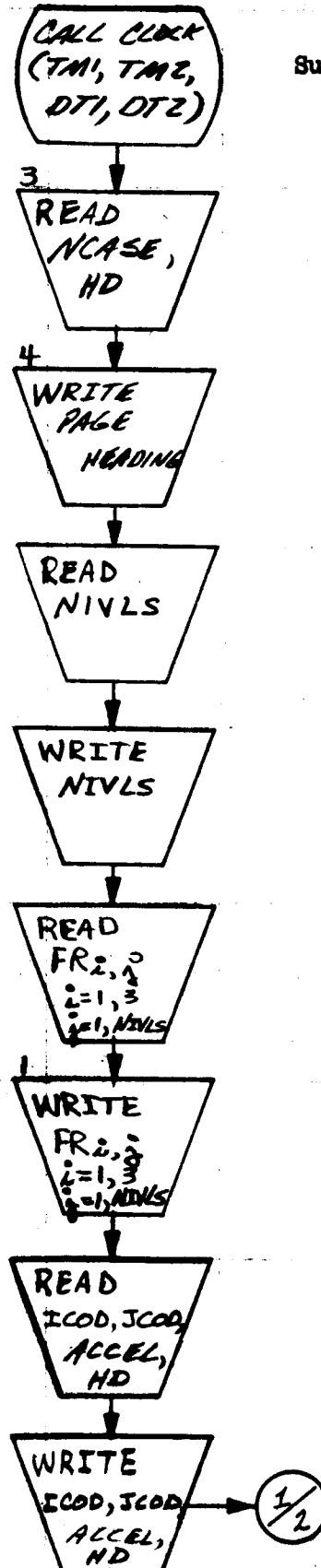


Figure 4.6.6-1

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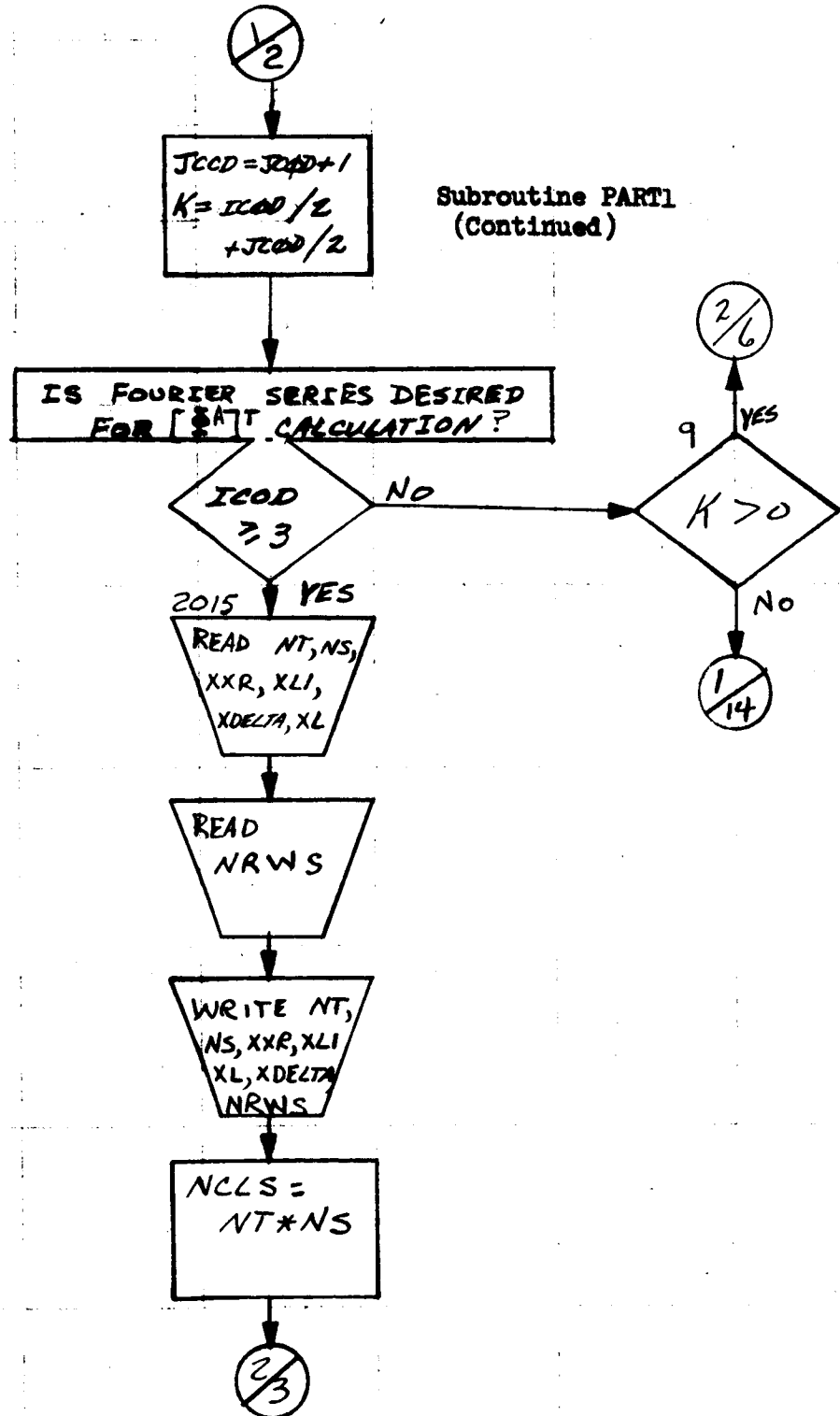


Figure 4.6.6-1 (Continued)

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**Subroutine PART1
(Continued)**

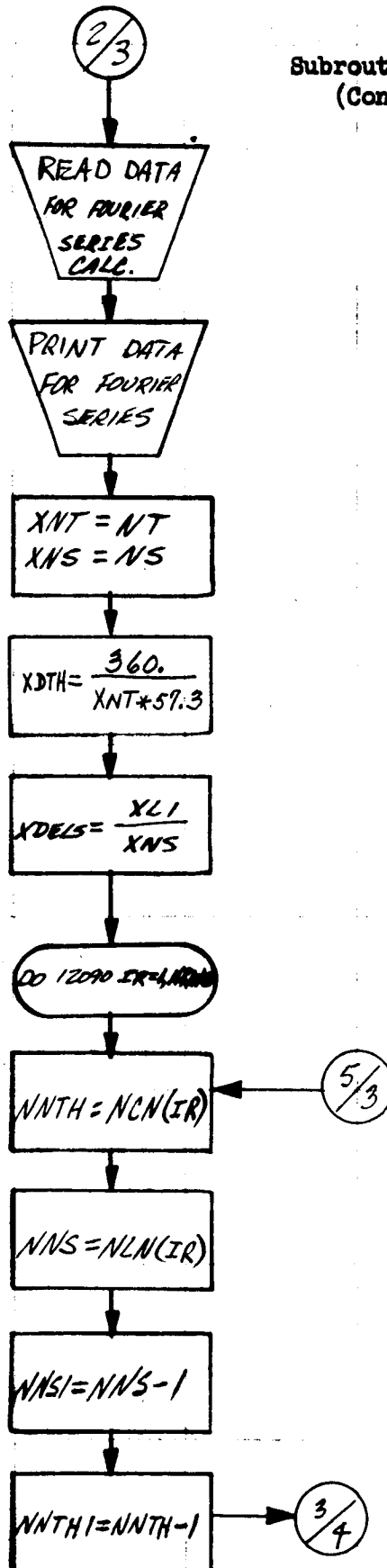
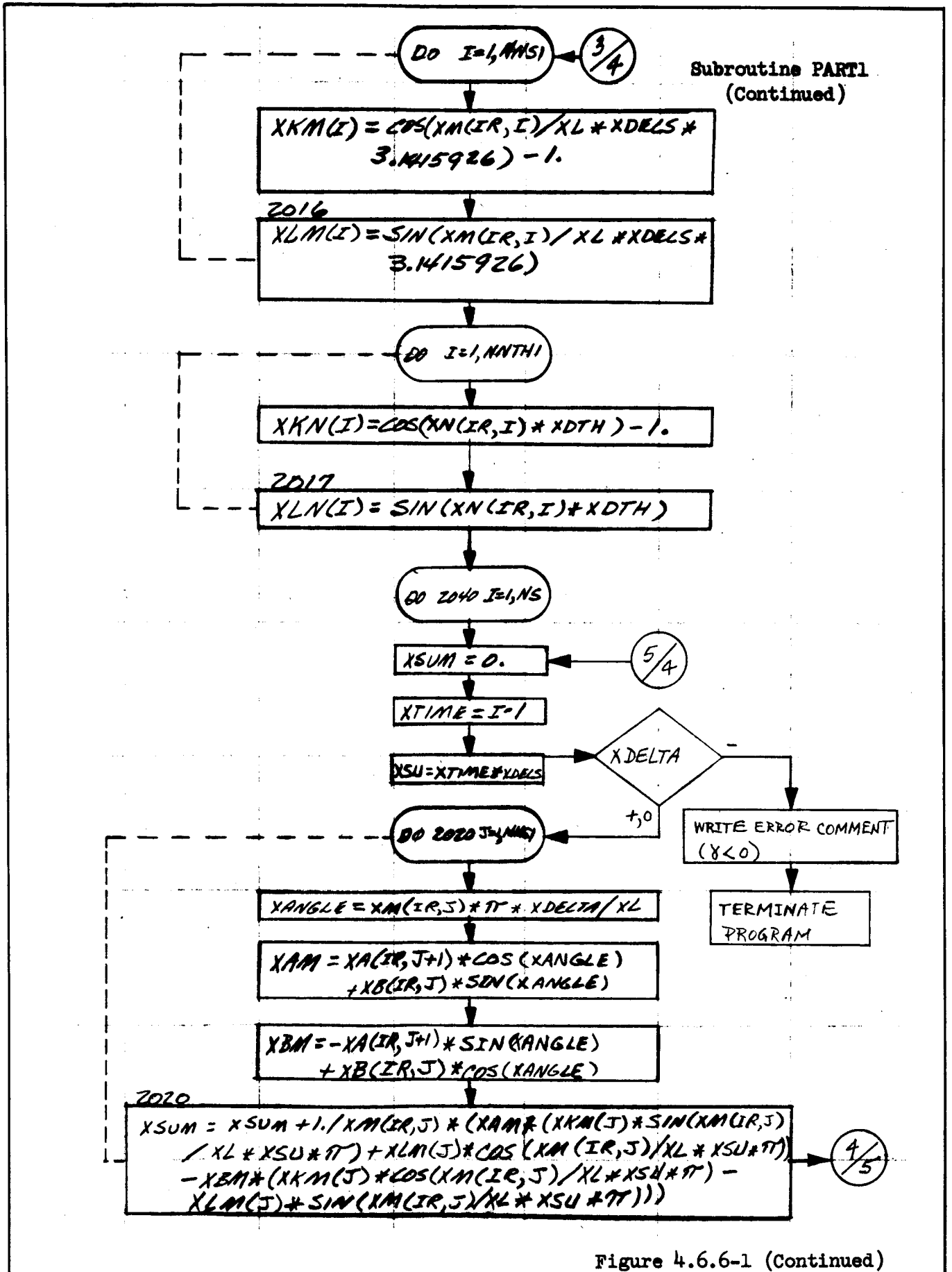


Figure 4.6.6-1 (Continued)



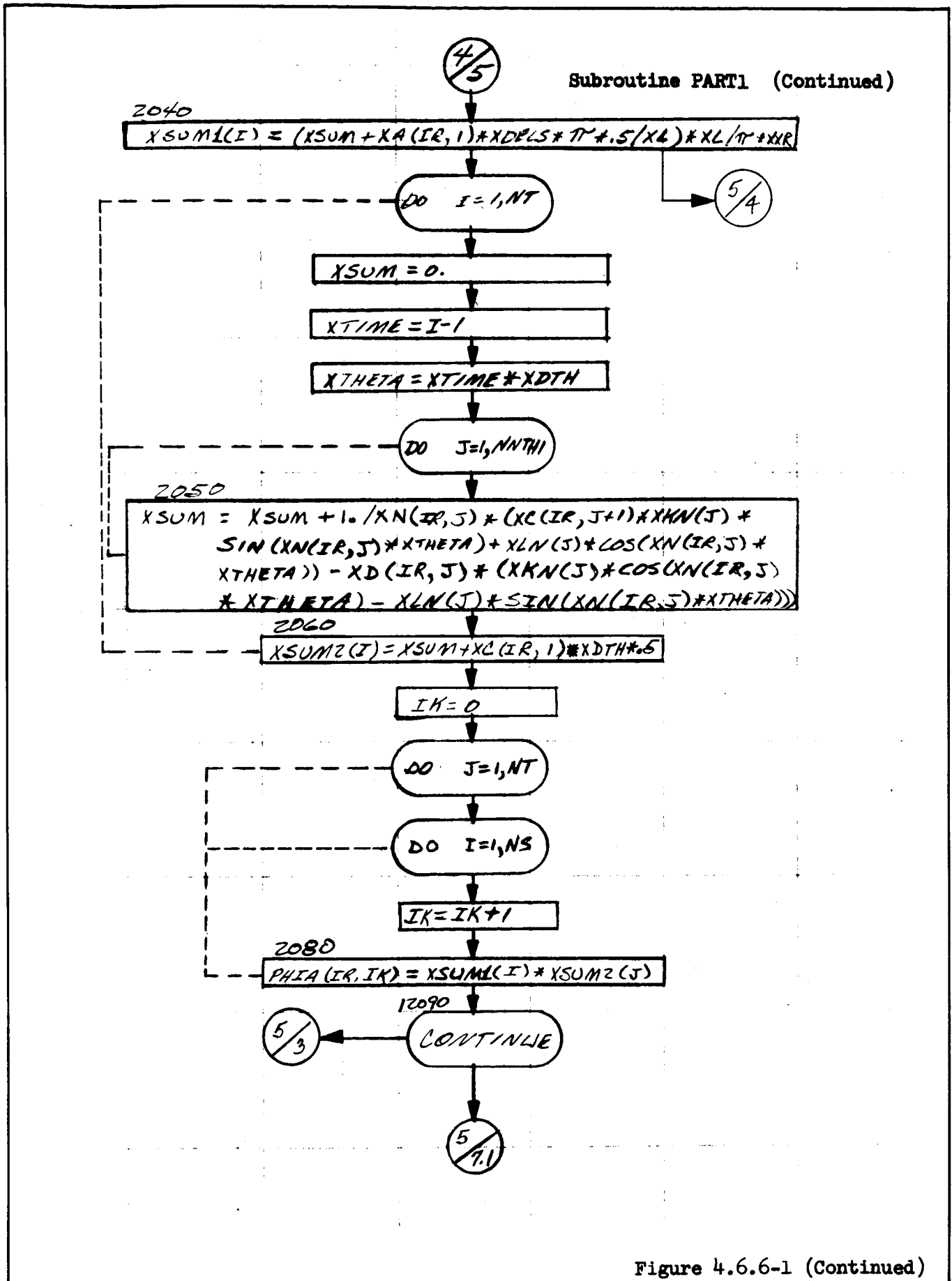


Figure 4.6.6-1 (Continued)

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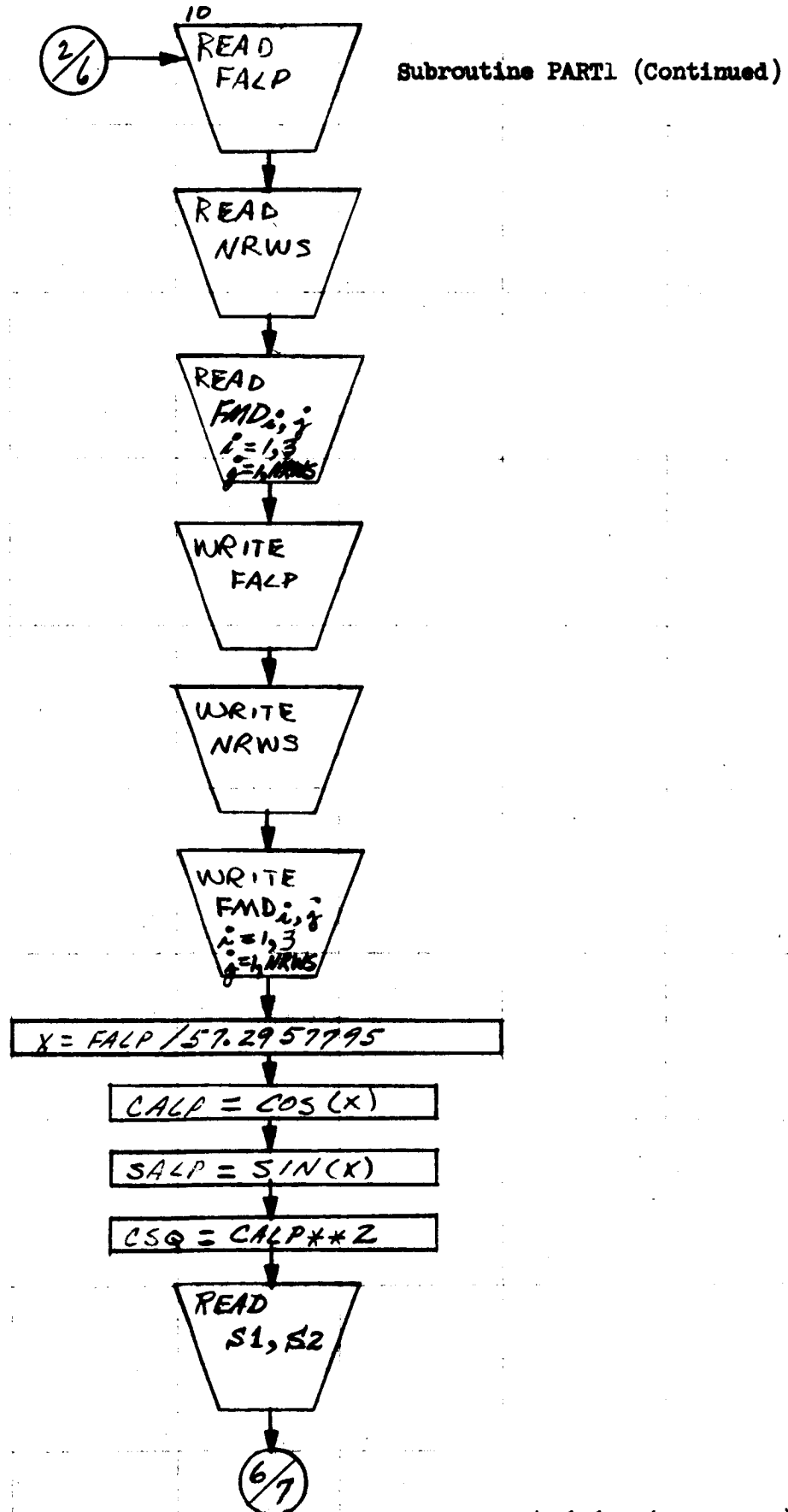


Figure 4.6.6-1 (Continued)

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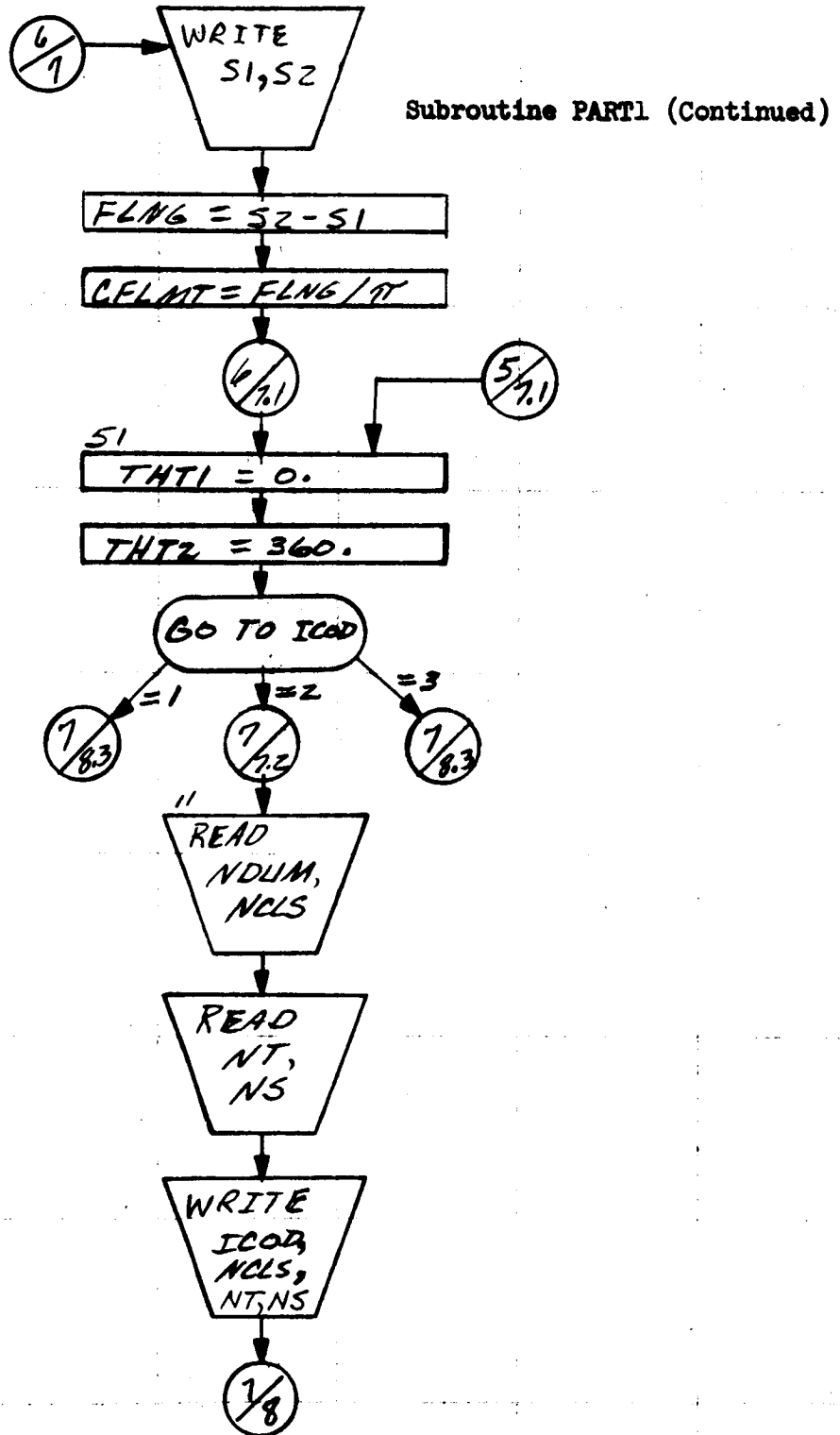


Figure 4.6.6-1 (Continued)

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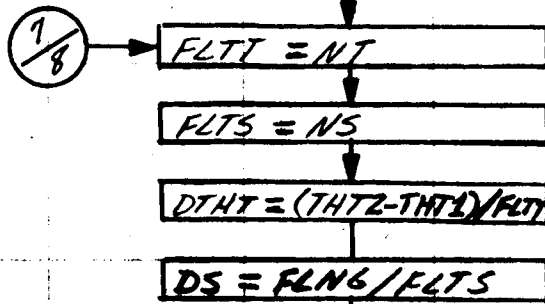
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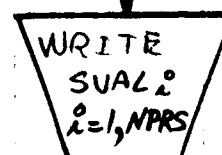
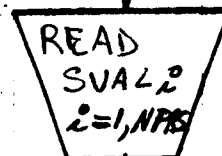
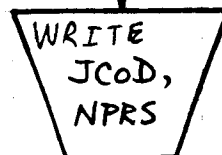
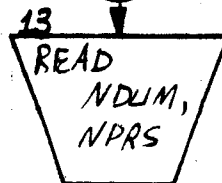
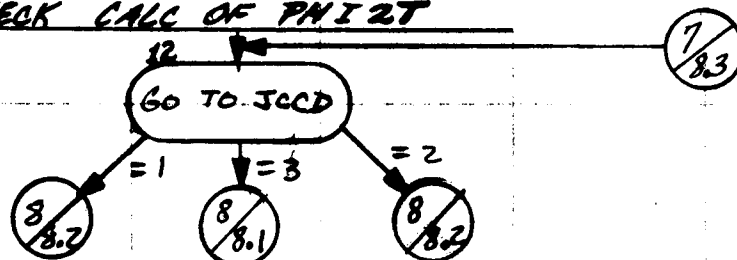
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CALC BASIC PARAMETERS



Subroutine PART1
(Continued)

CHECK CALC OF PH12T



CYCLE ON MODE COUNTER

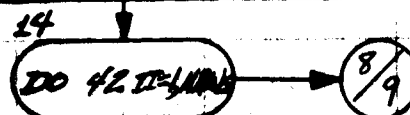


Figure 4.6.6-1 (Continued)

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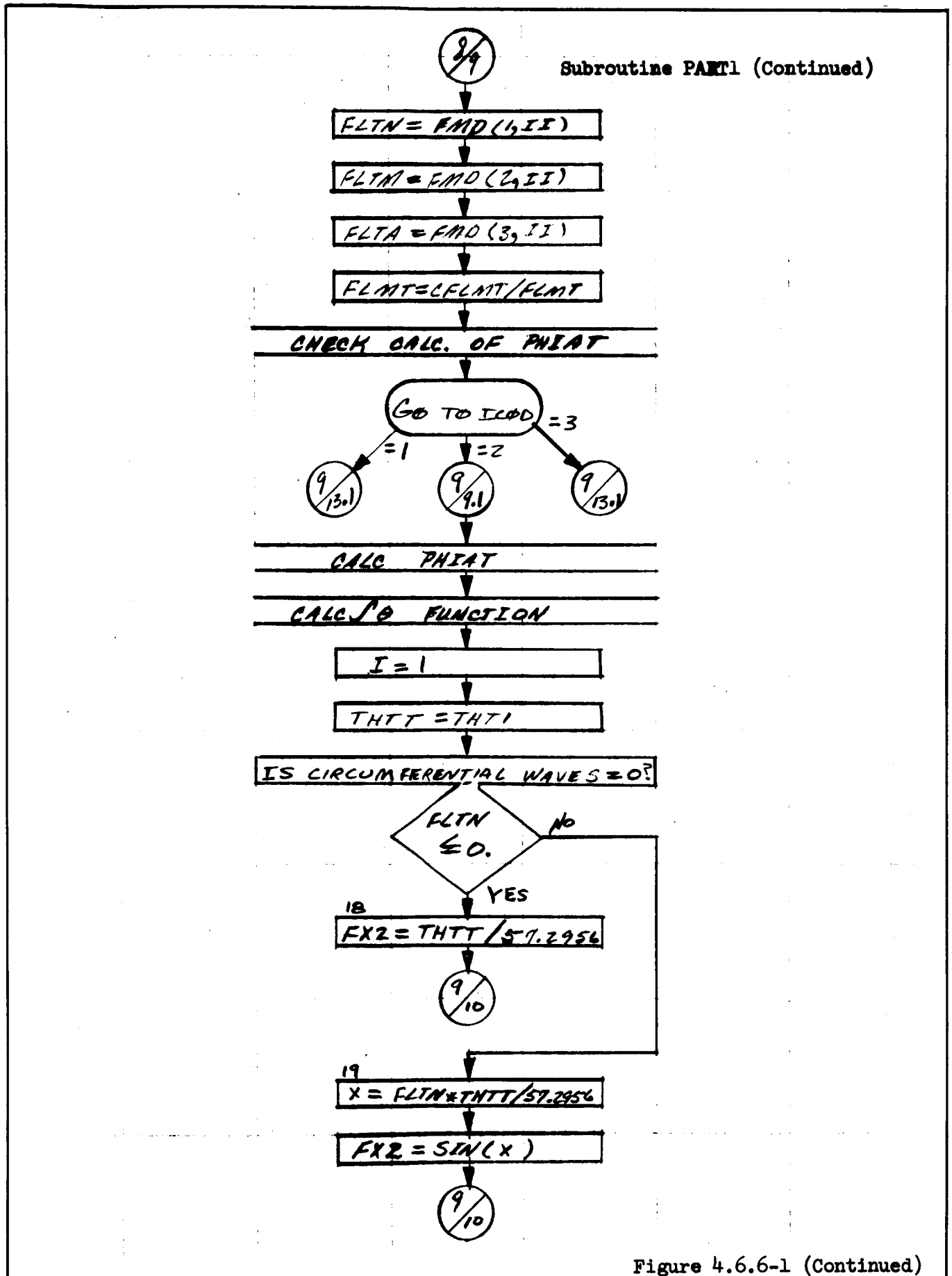


Figure 4.6.6-1 (Continued)

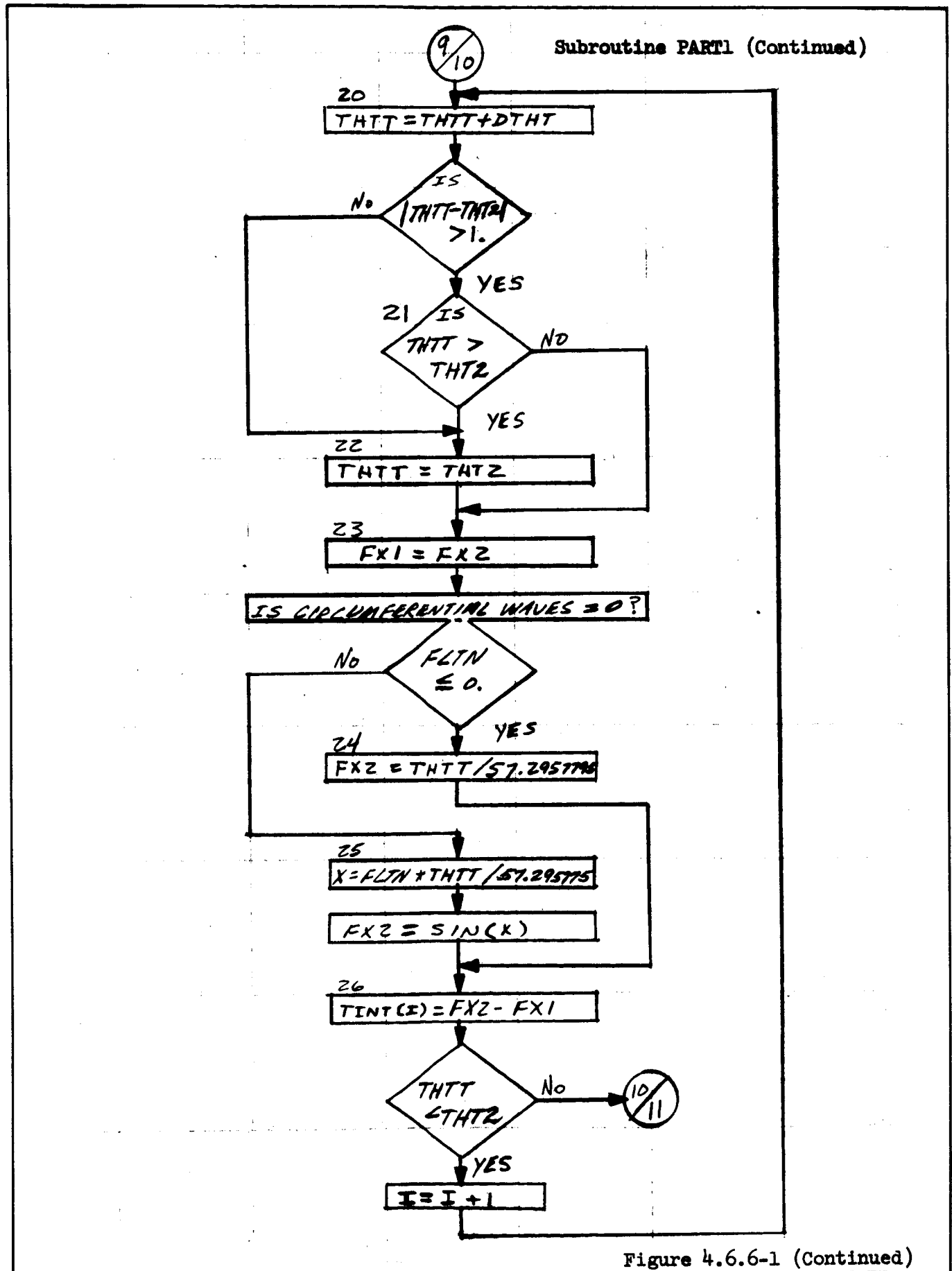


Figure 4.6.6-1 (Continued)

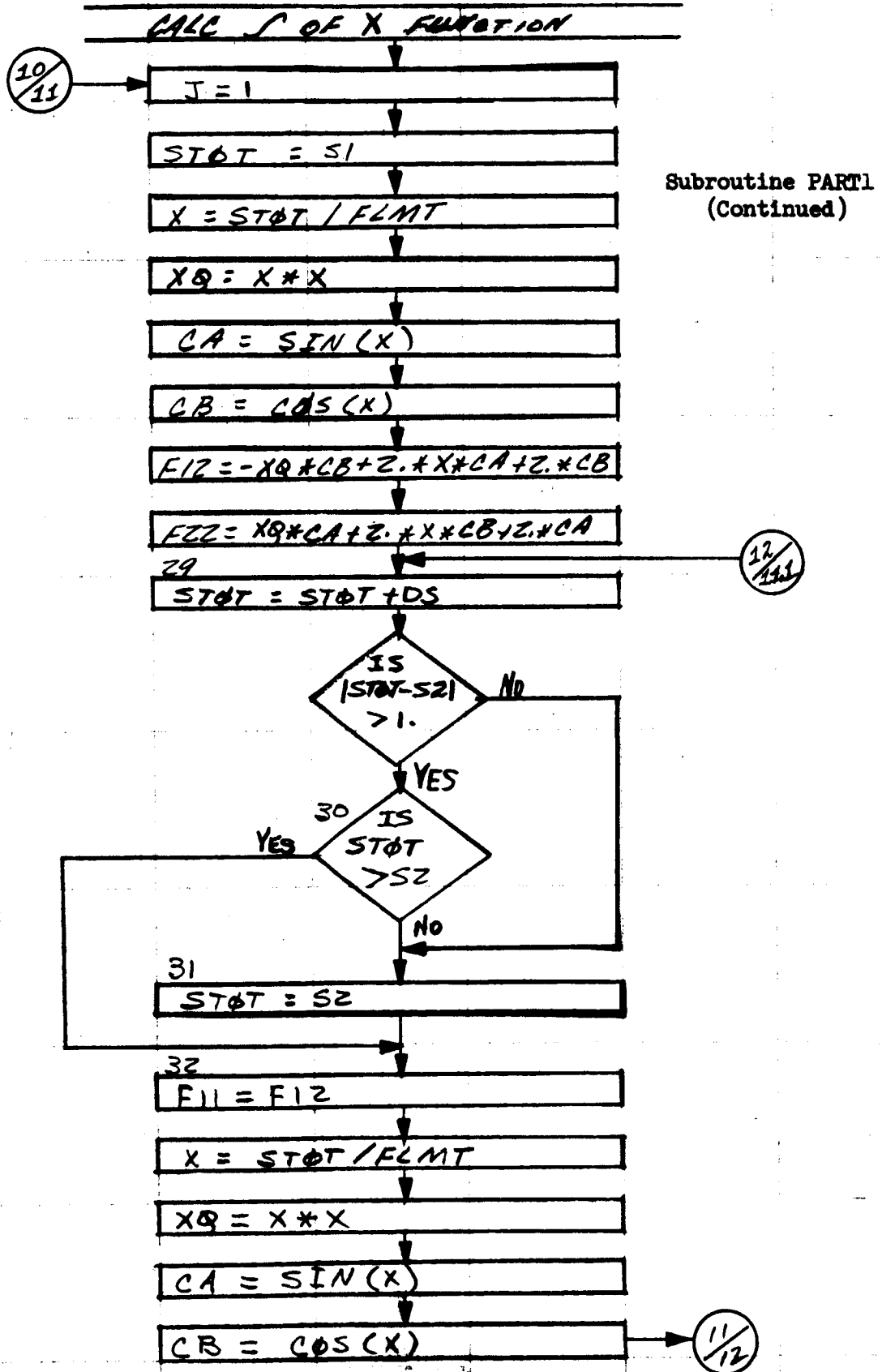


Figure 4.6.6-1 (Continued)

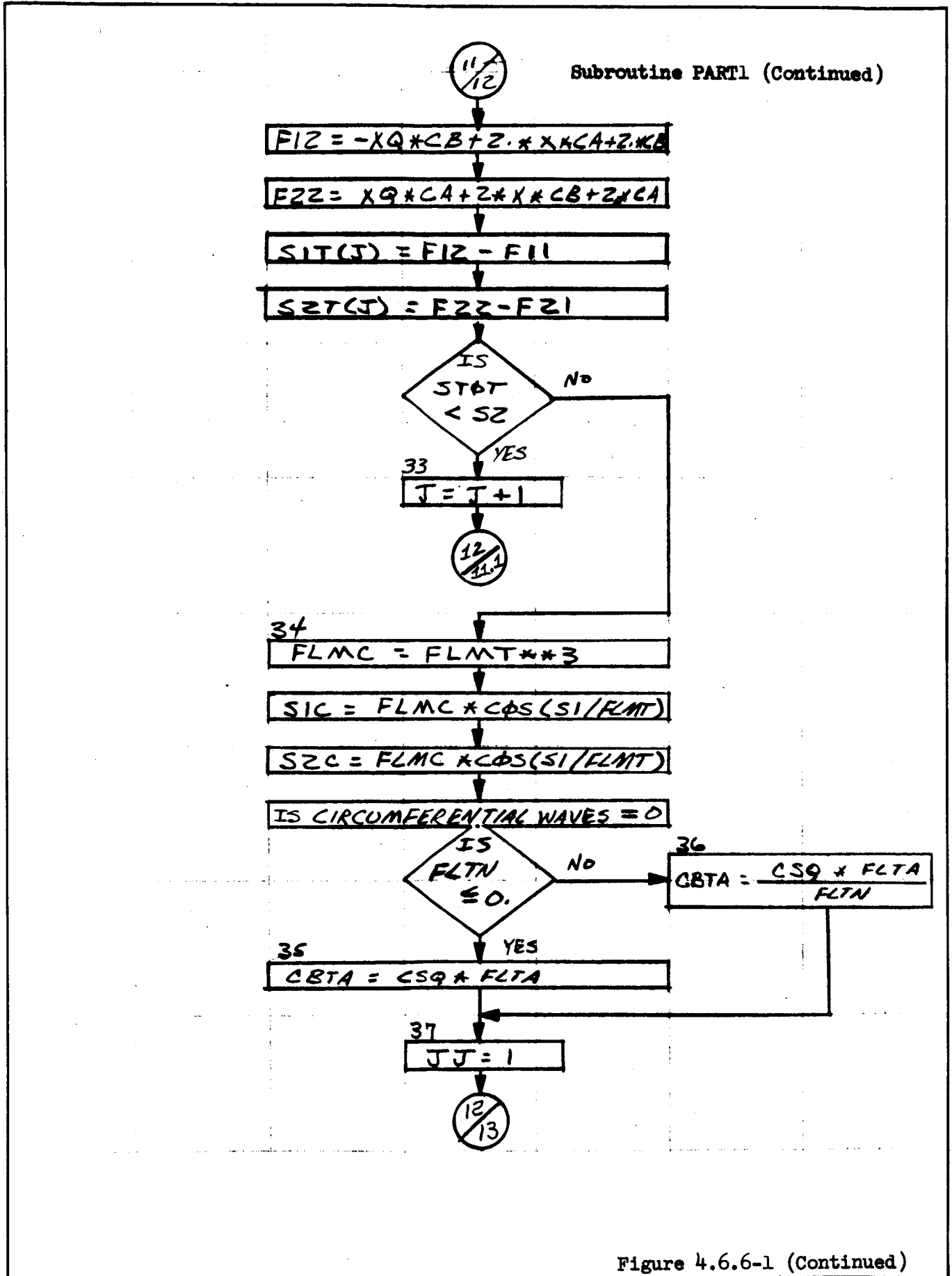


Figure 4.6.6-1 (Continued)

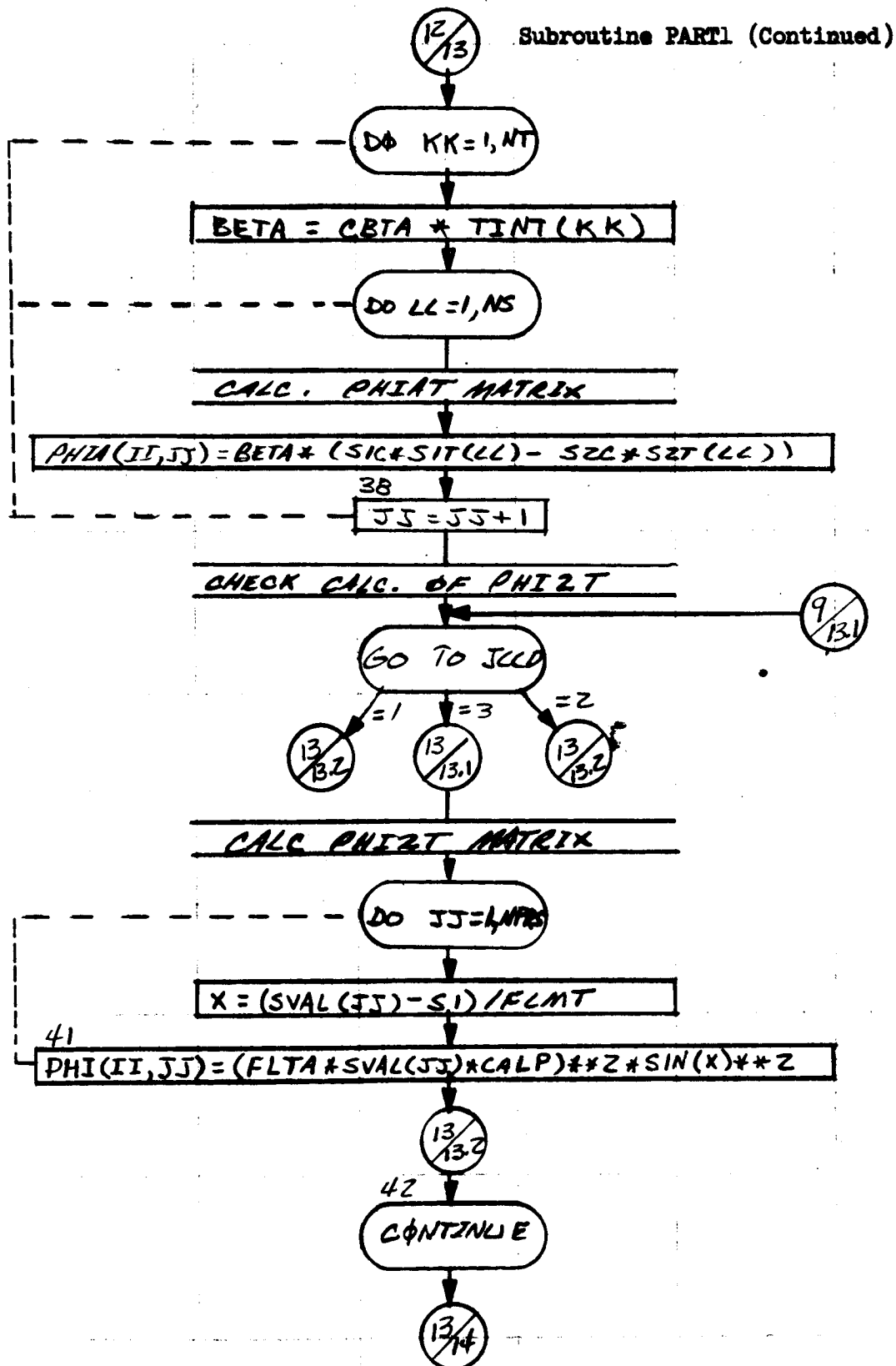


Figure 4.6.6-1 (Continued)

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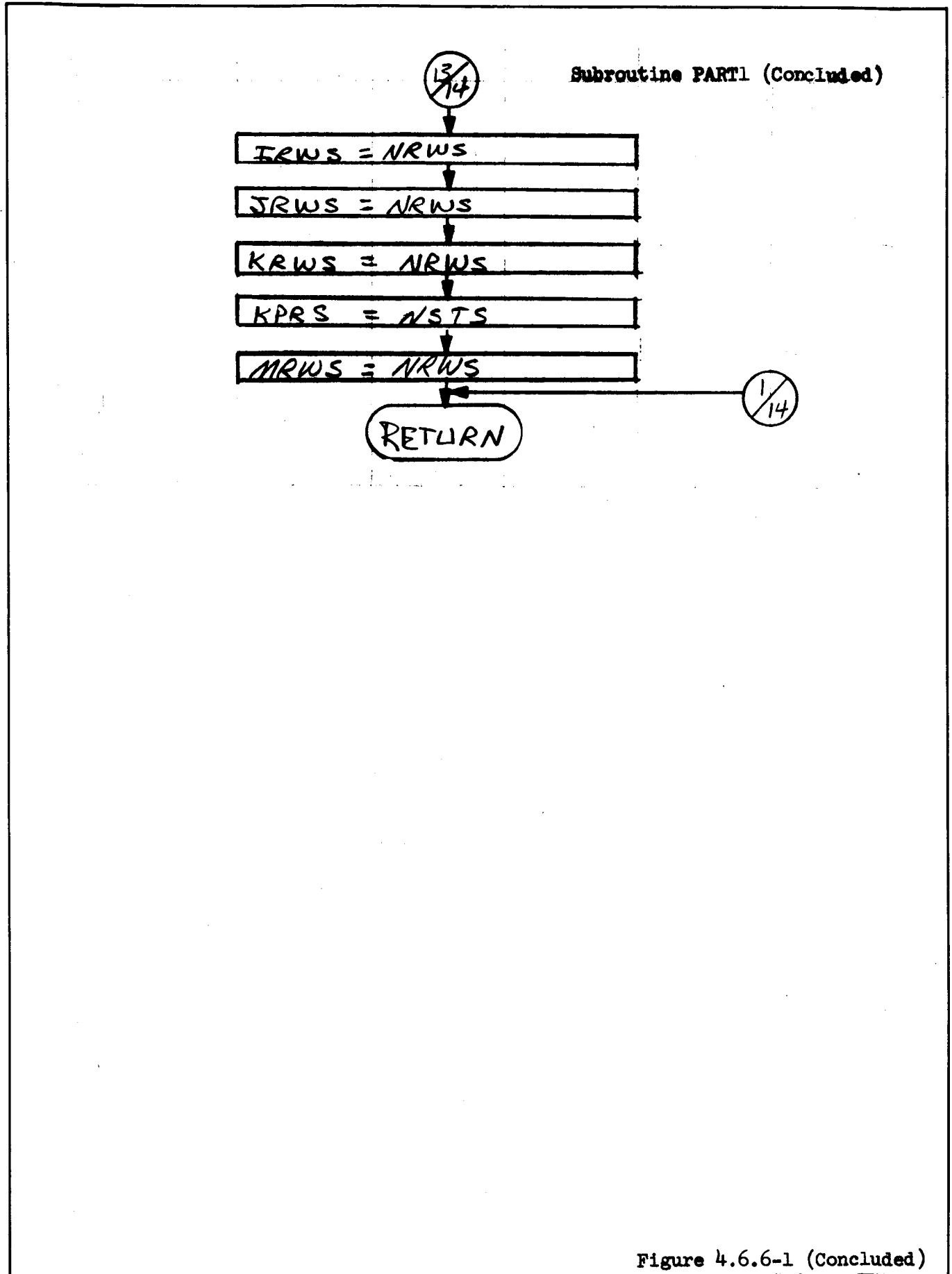


Figure 4.6.6-1 (Concluded)

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4.6.7 Link Two, FHALF. - Figure 4.6.7-1 shows the remaining input portion for each case. It consists of 14 pages. Control returns to the MAIN program after this link is completed.

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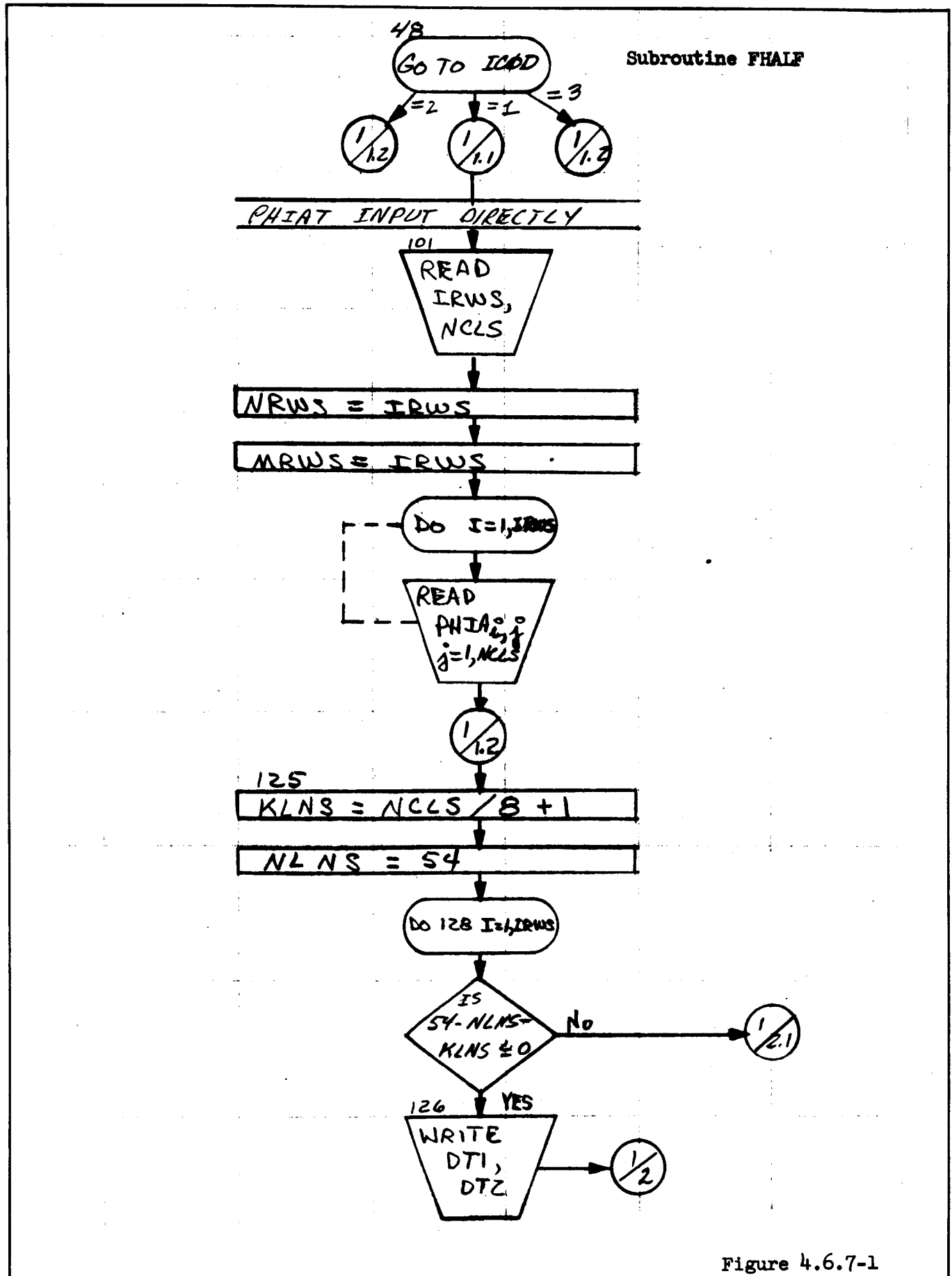


Figure 4.6.7-1

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Subroutine PHALF (Continued)

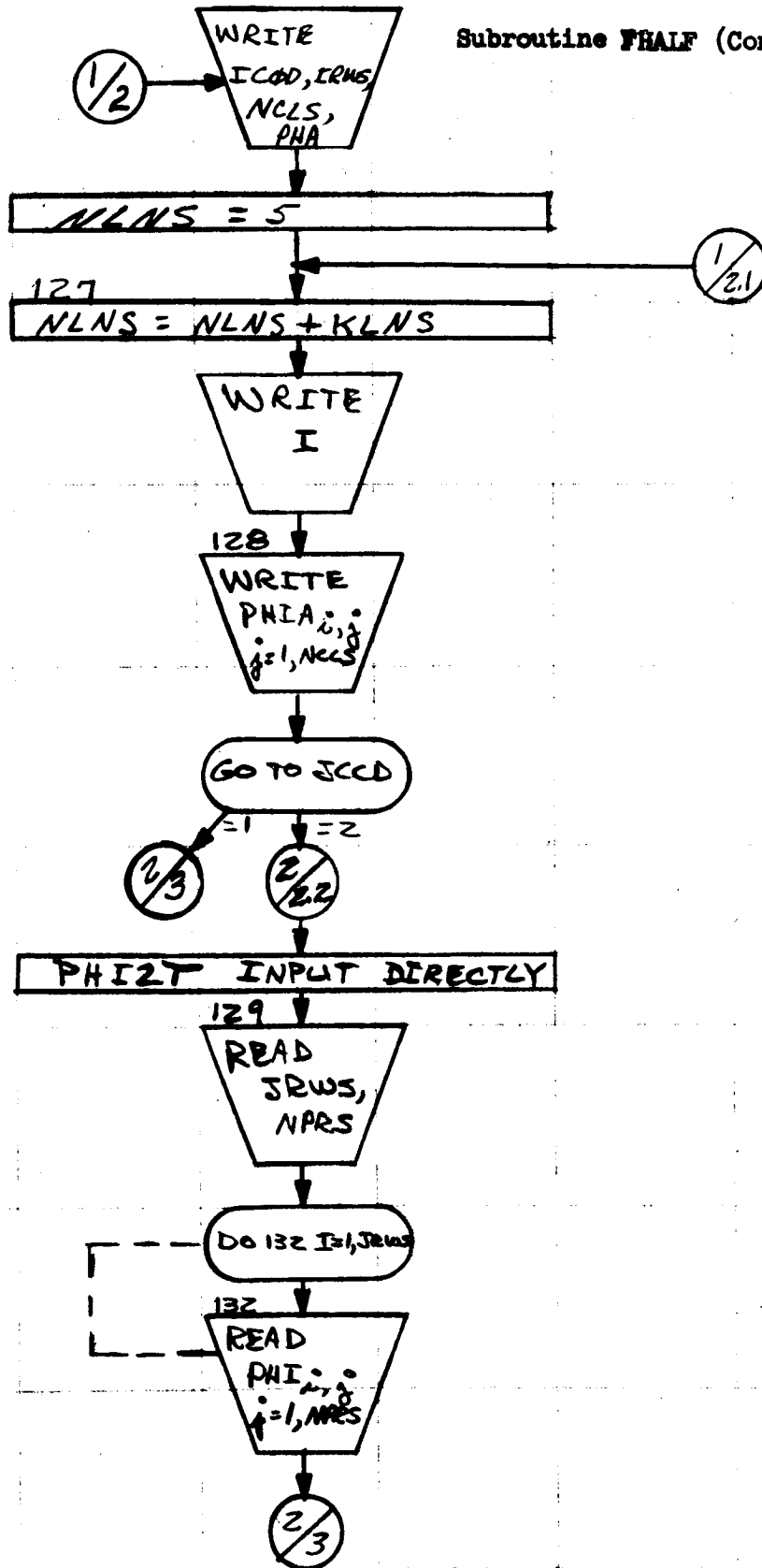


Figure 4.6.7-1 (Continued)

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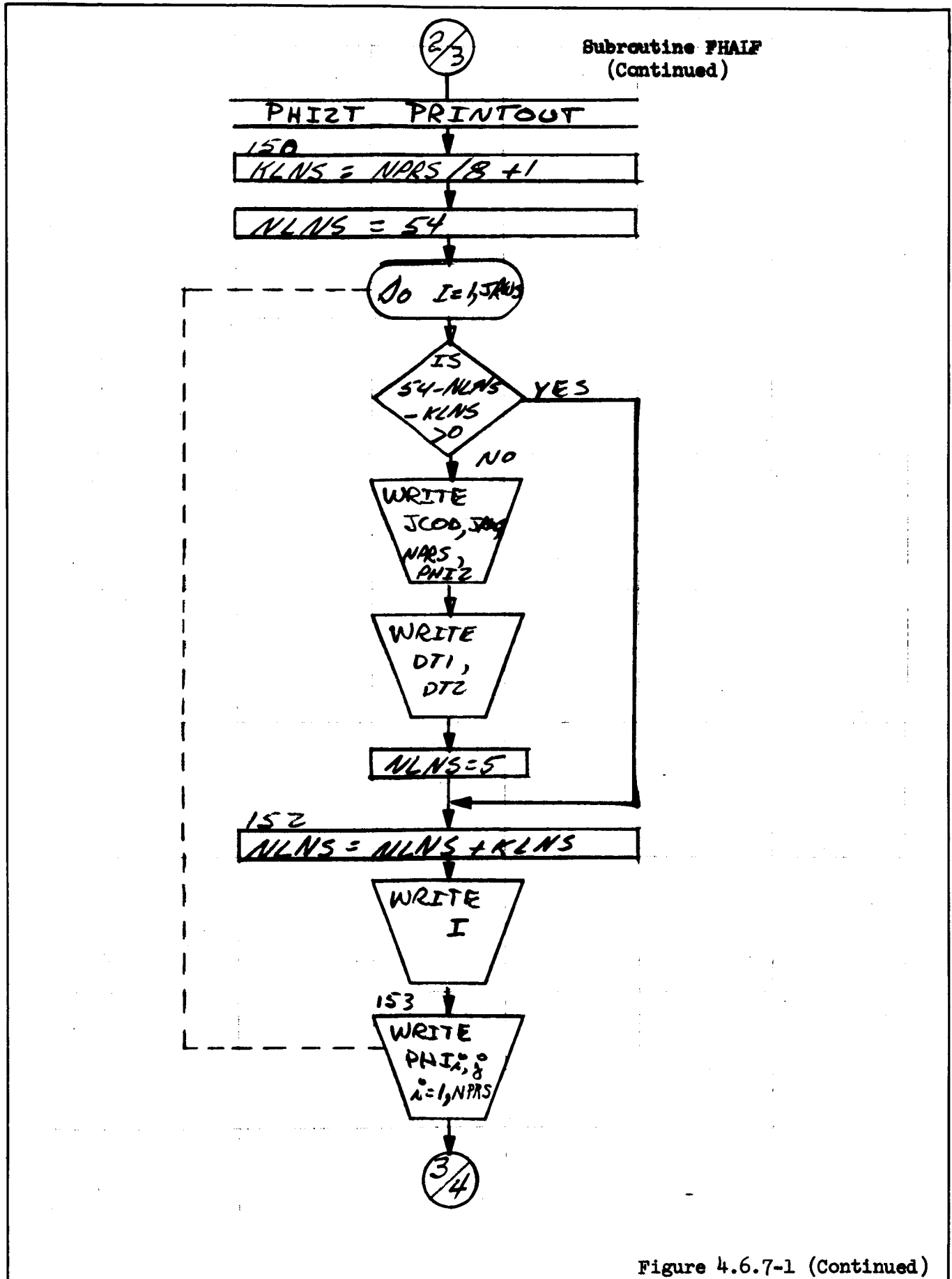


Figure 4.6.7-1 (Continued)

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Subroutine PHALF (Continued)

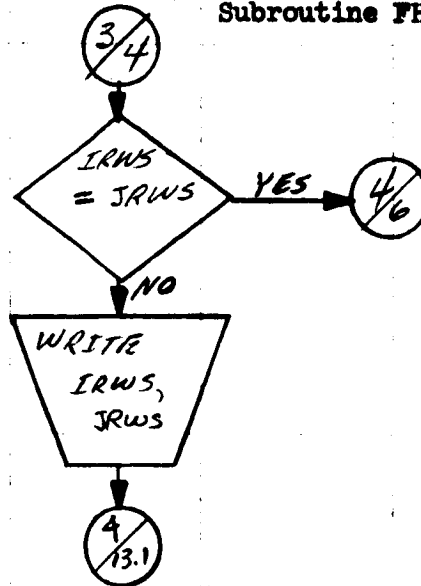


Figure 4.6.7-1 (Continued)

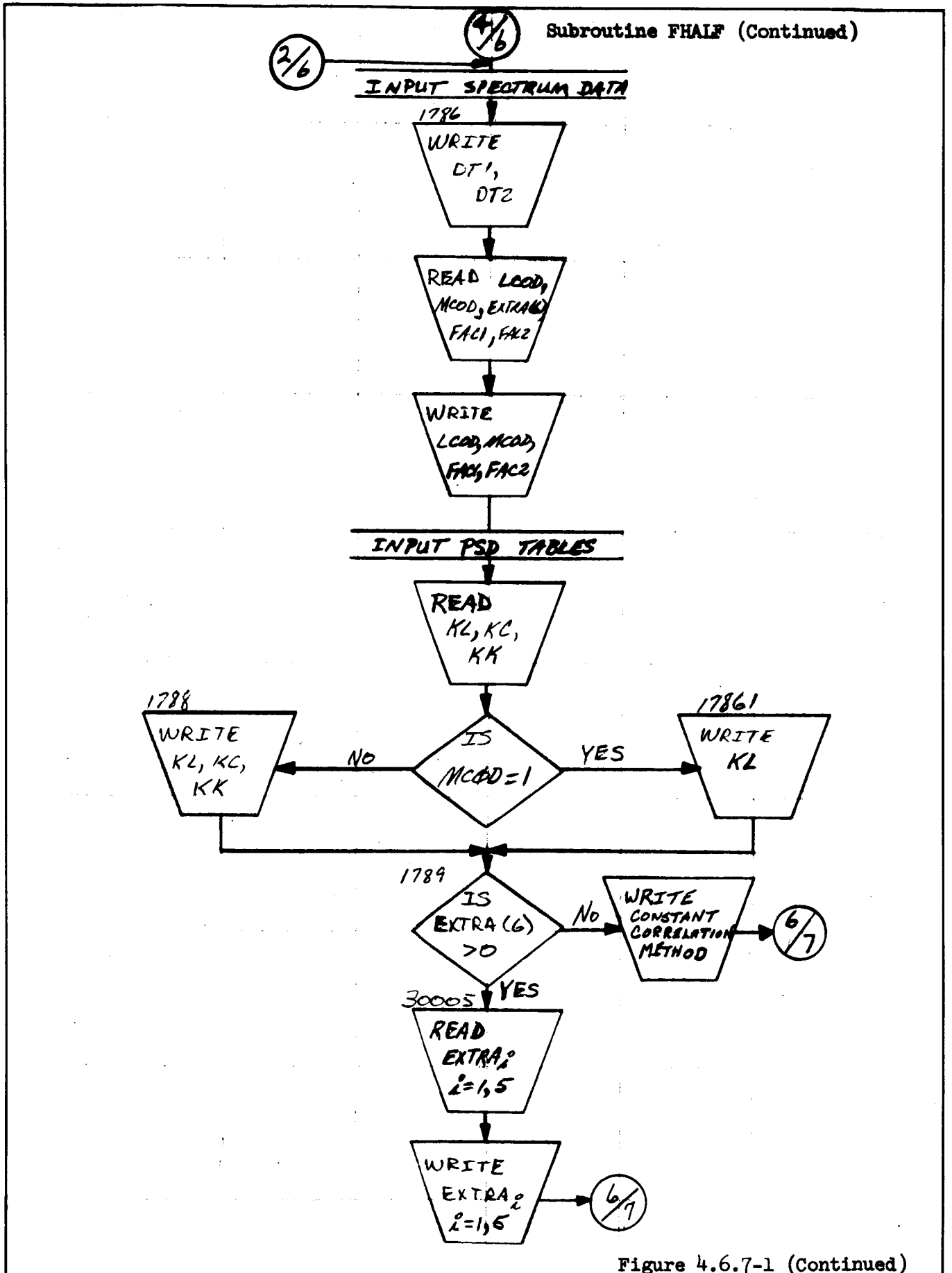


Figure 4.6.7-1 (Continued)

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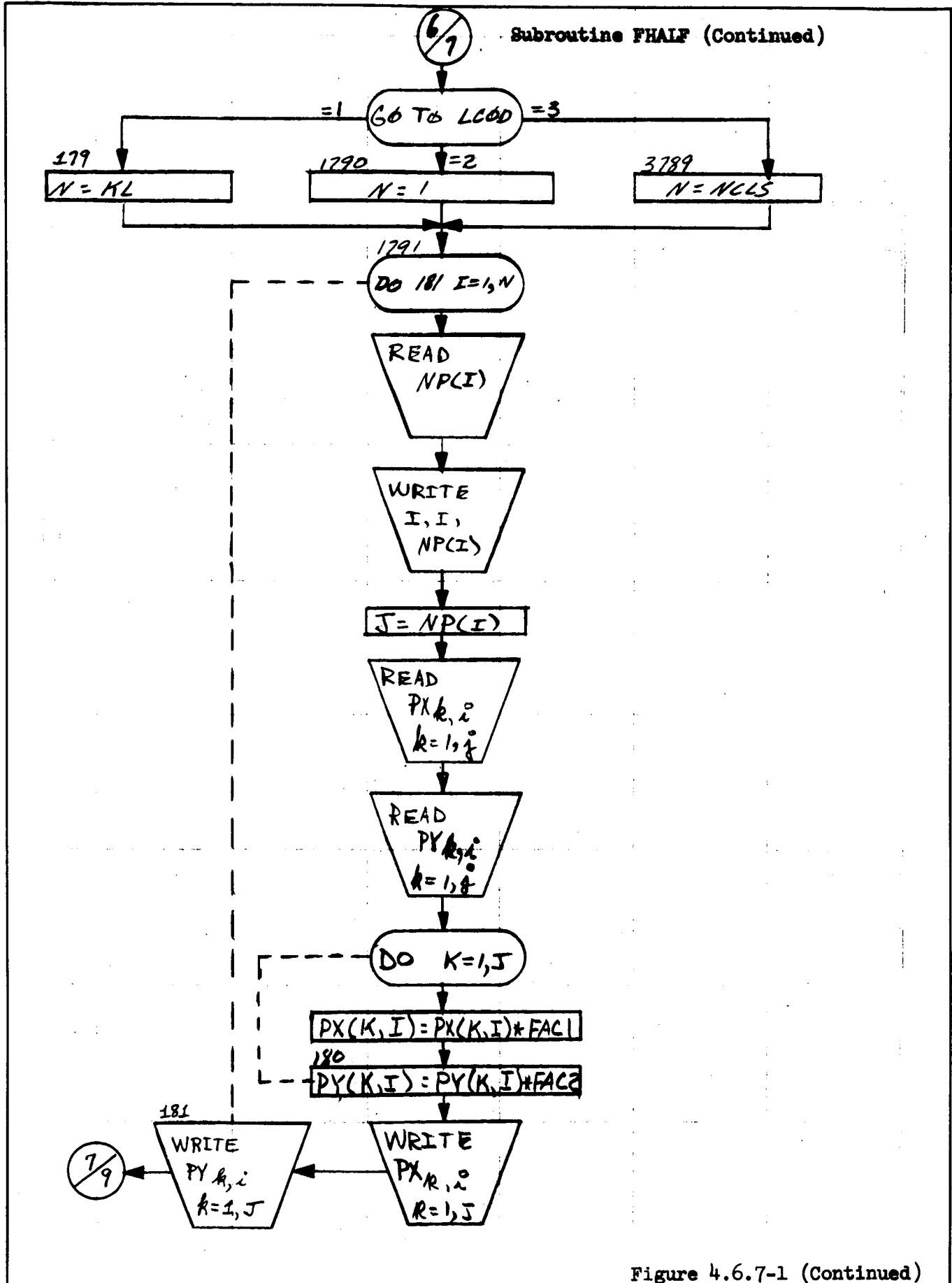


Figure 4.6.7-1 (Continued)

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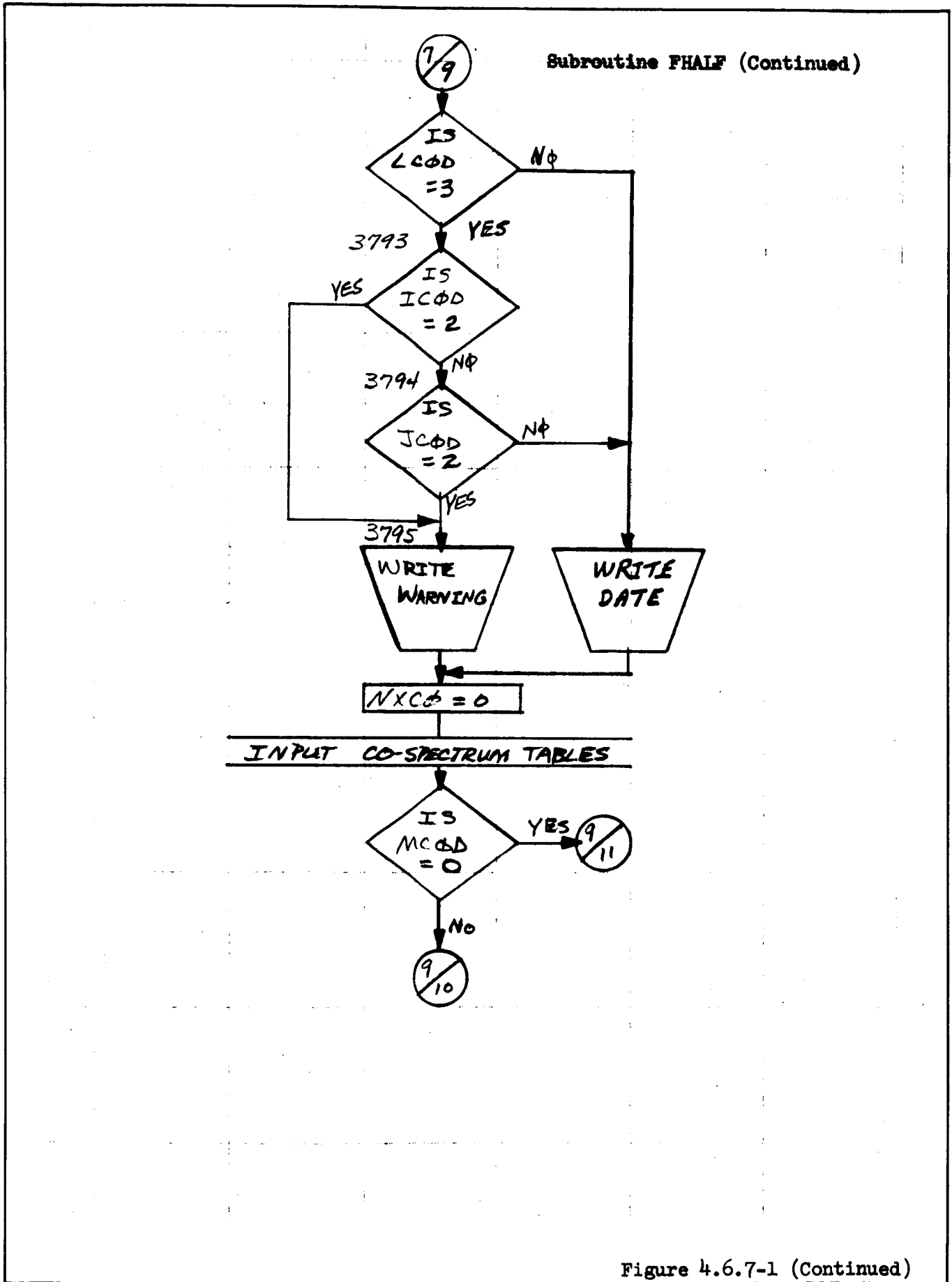


Figure 4.6.7-1 (Continued)

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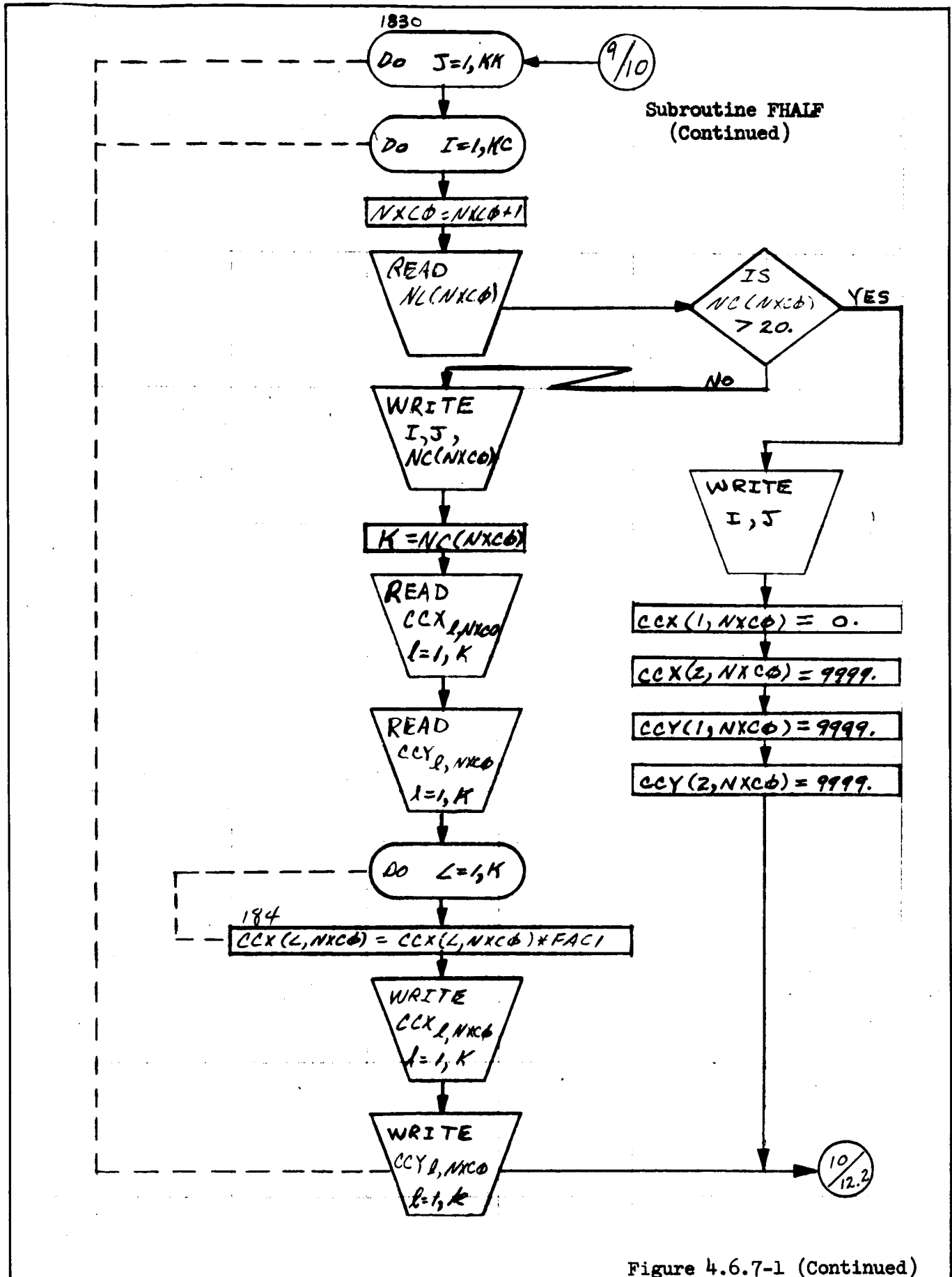


Figure 4.6.7-1 (Continued)

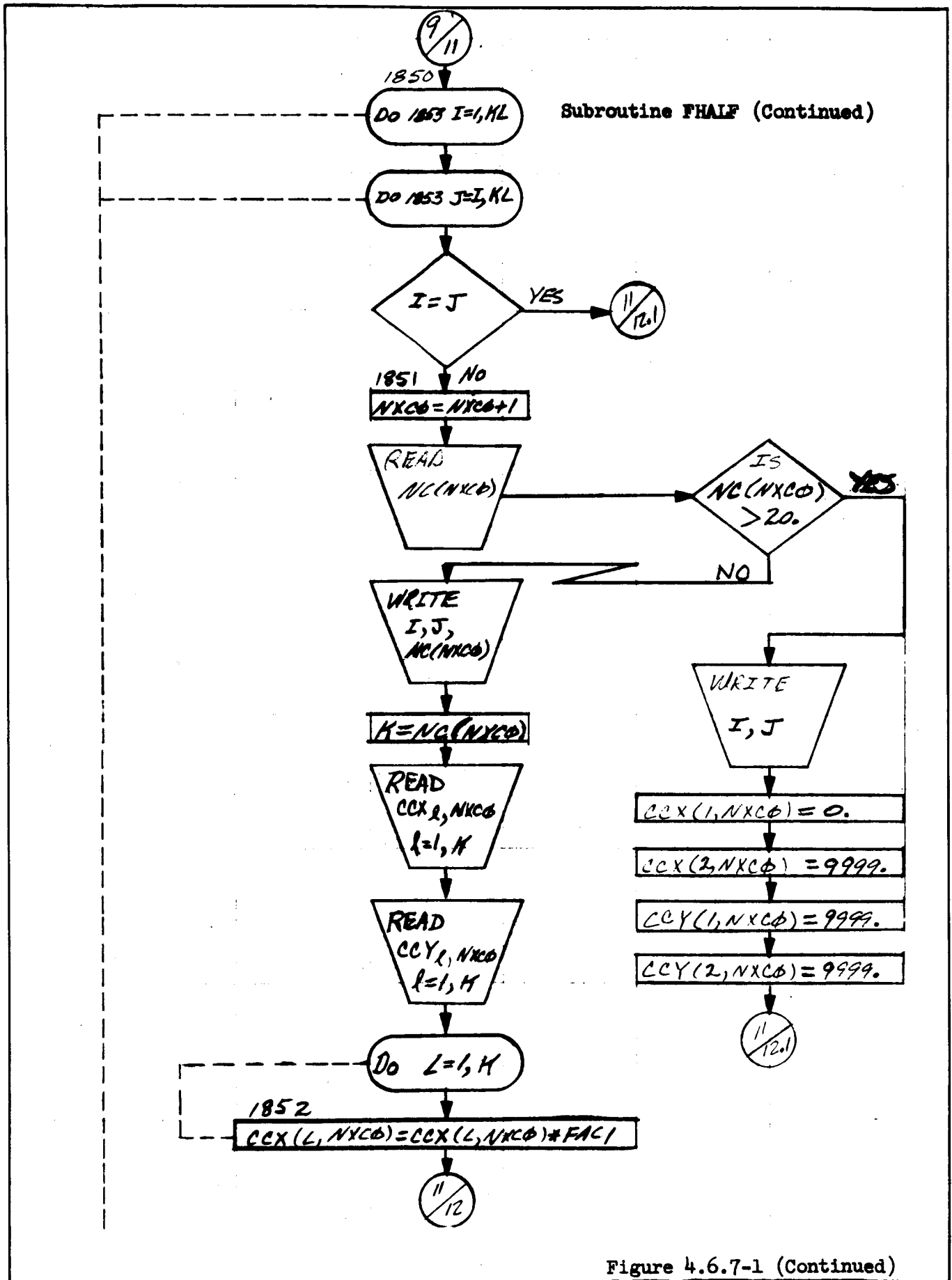


Figure 4.6.7-1 (Continued)

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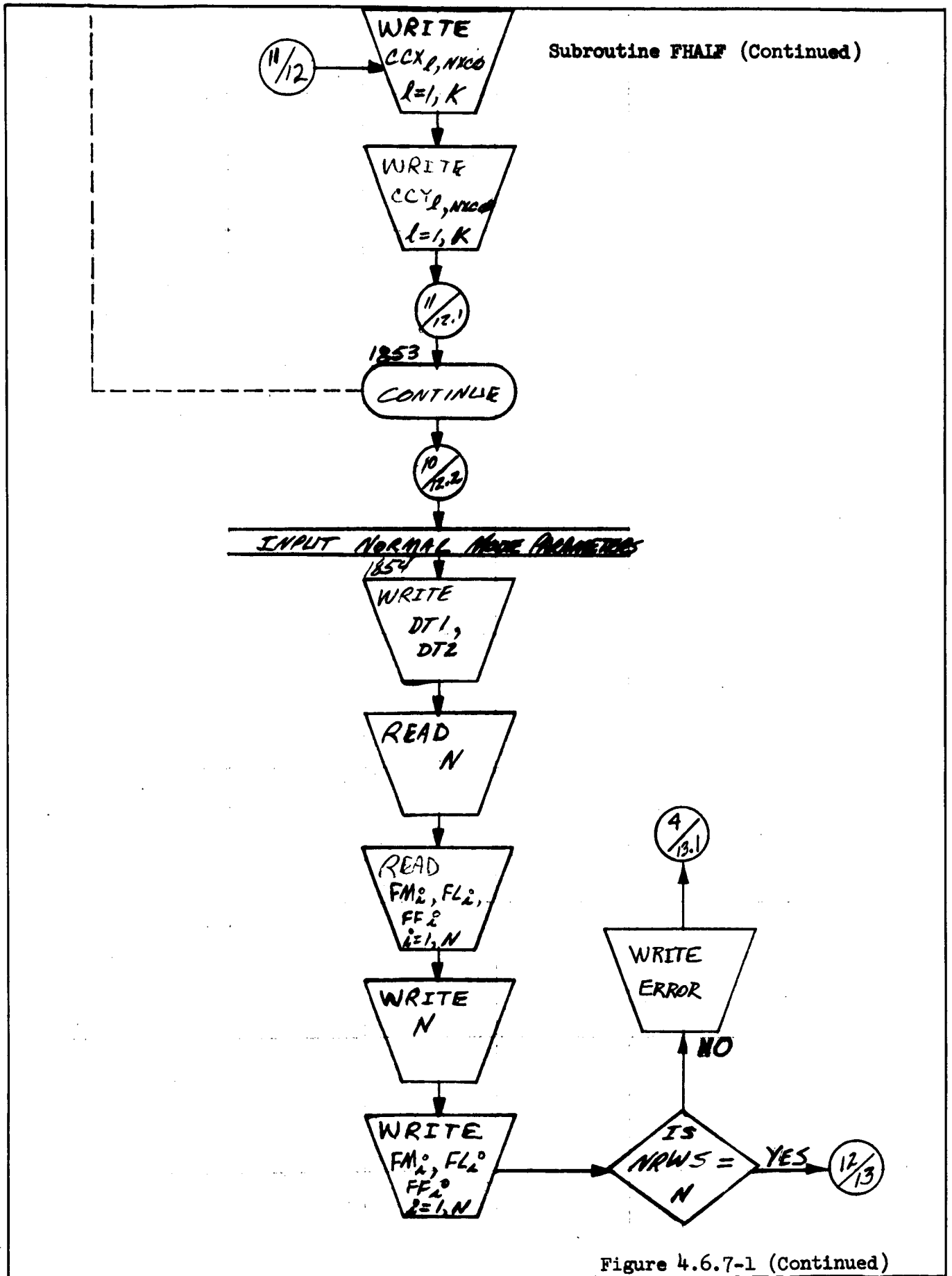


Figure 4.6.7-1 (Continued)

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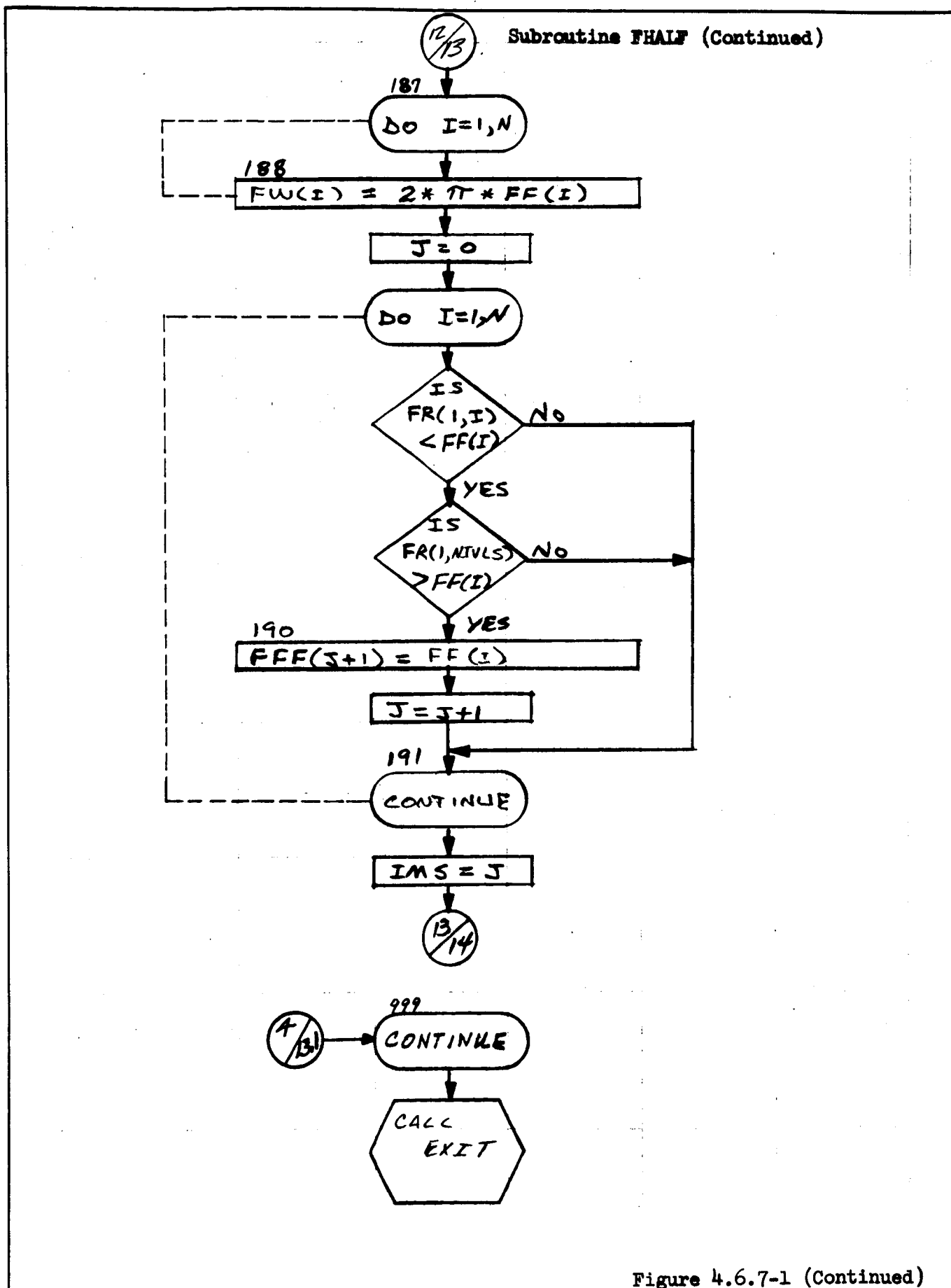


Figure 4.6.7-1 (Continued)

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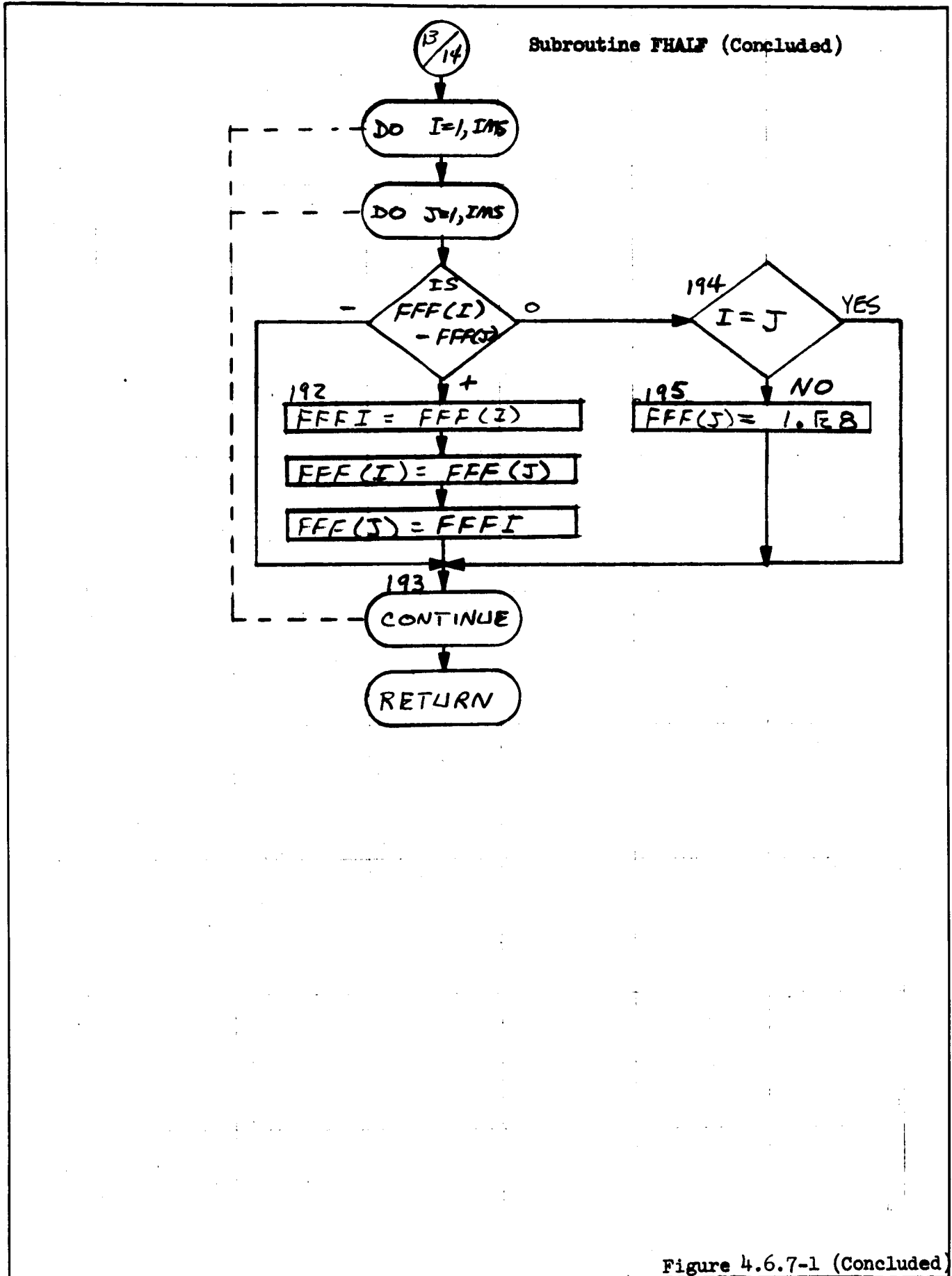


Figure 4.6.7-1 (Concluded)

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4.6.8 Link Three, SHALF. - Figure 4.6.8-1 shows the main cycle of the program and how the desired responses are calculated. This portion consists of 20 pages. Control returns to the MAIN program after this link is completed to begin on the next case, if any.

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SET-UP CONSTANTS

IPS = 1

INS = 1

Subroutine SHALF

INITIALIZE FREQUENCY CYCLE

DO I=1, MINS

$R(I) = 0.$

$R1(I) = 0.$

$T(I) = 0.$

$TT(I) = 0.$

$NVAL = 0.$

$NIVL = 1$

$FREQ = FR(1, NIVL)$

$DEFQ = FR(2, NIVL)$

$KCNT = FR(3, NIVL) + .0000001$

$KPRC = KCNT$

$X1 = EXTRA(1) * EXTRA(3)$

$X2 = EXTRA(2) * EXTRA(4)$

1/2

Figure 4.6.8-1

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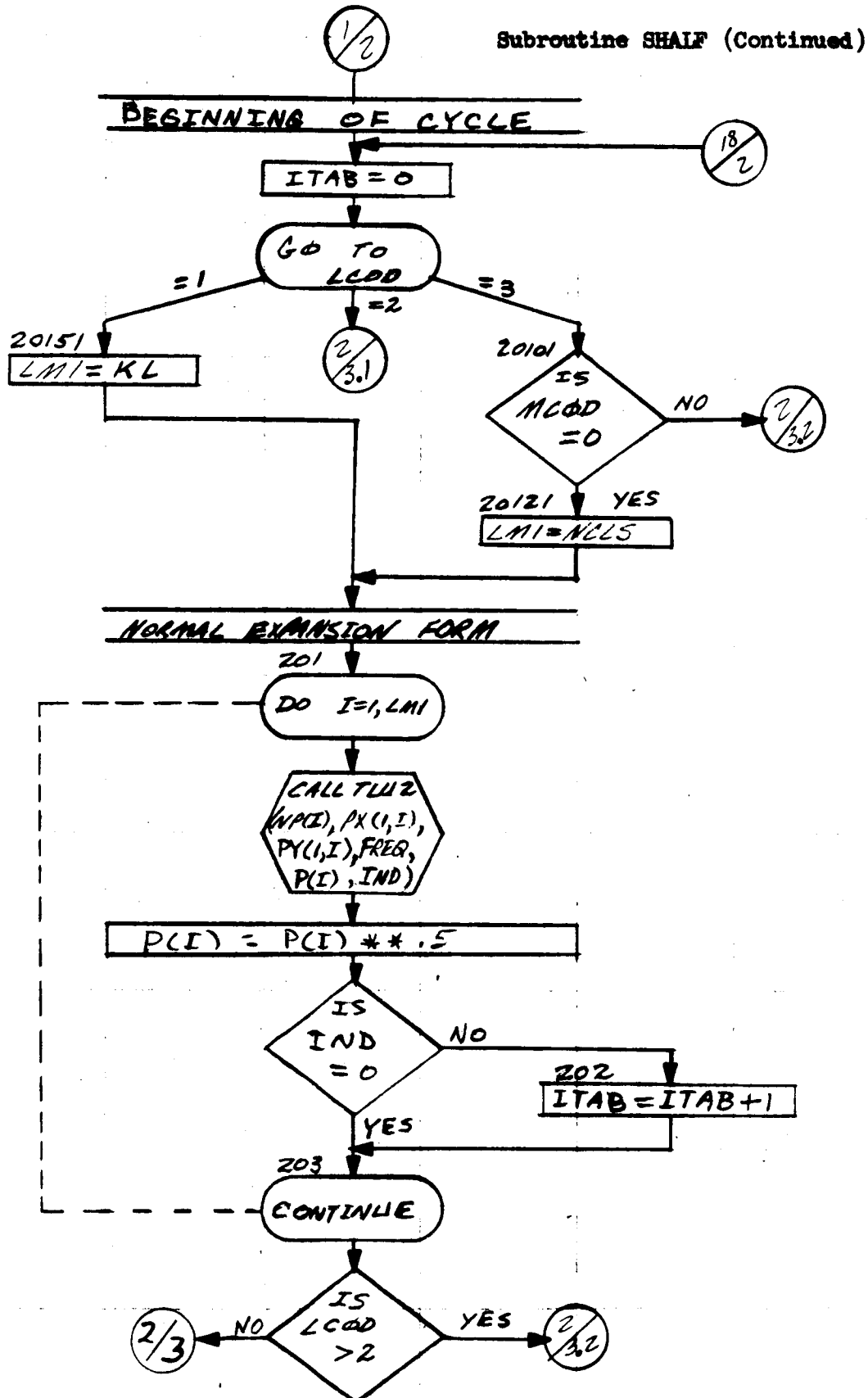


Figure 4.6.8-1 (Continued)

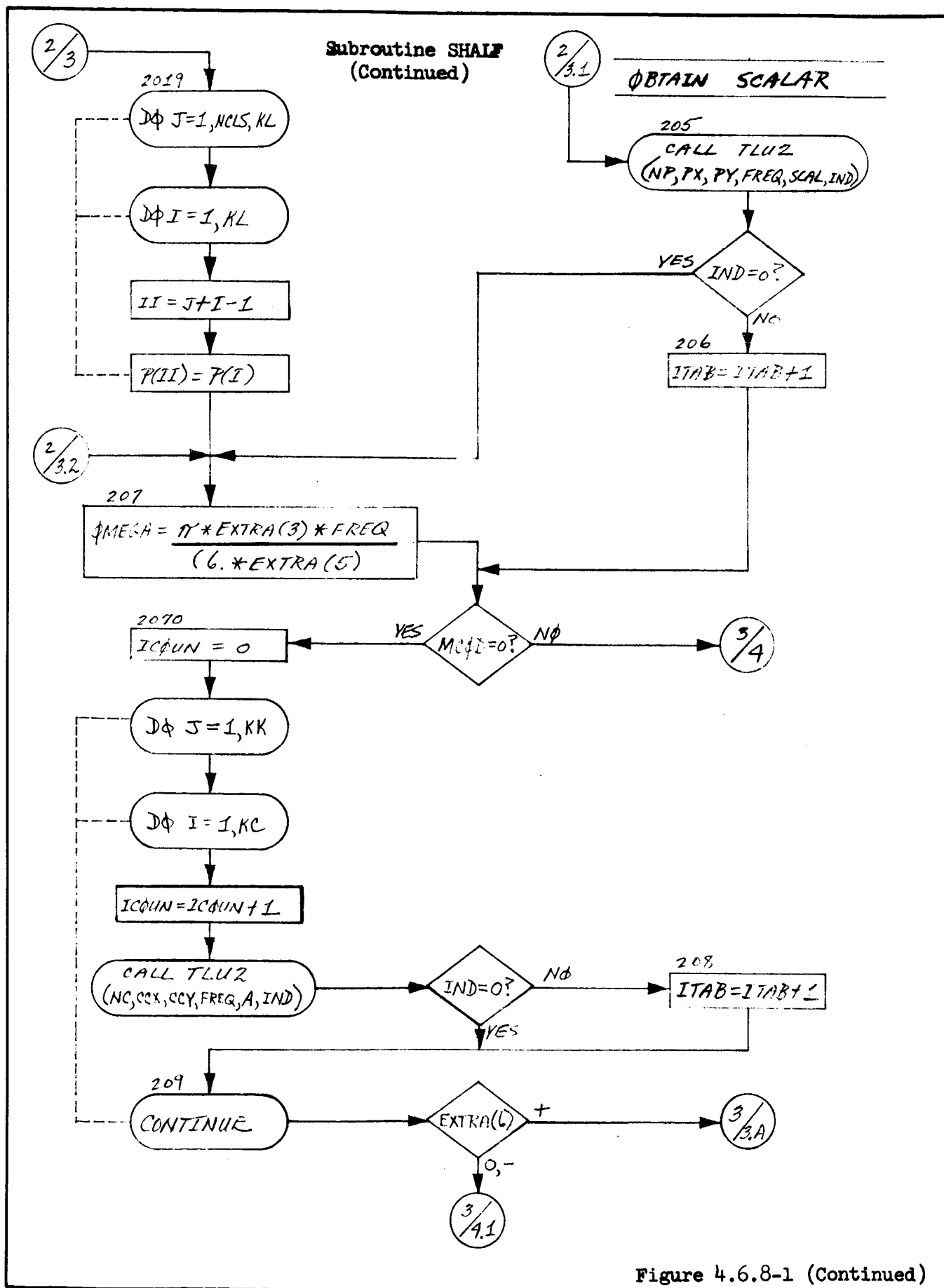


Figure 4.6.8-1 (Continued)

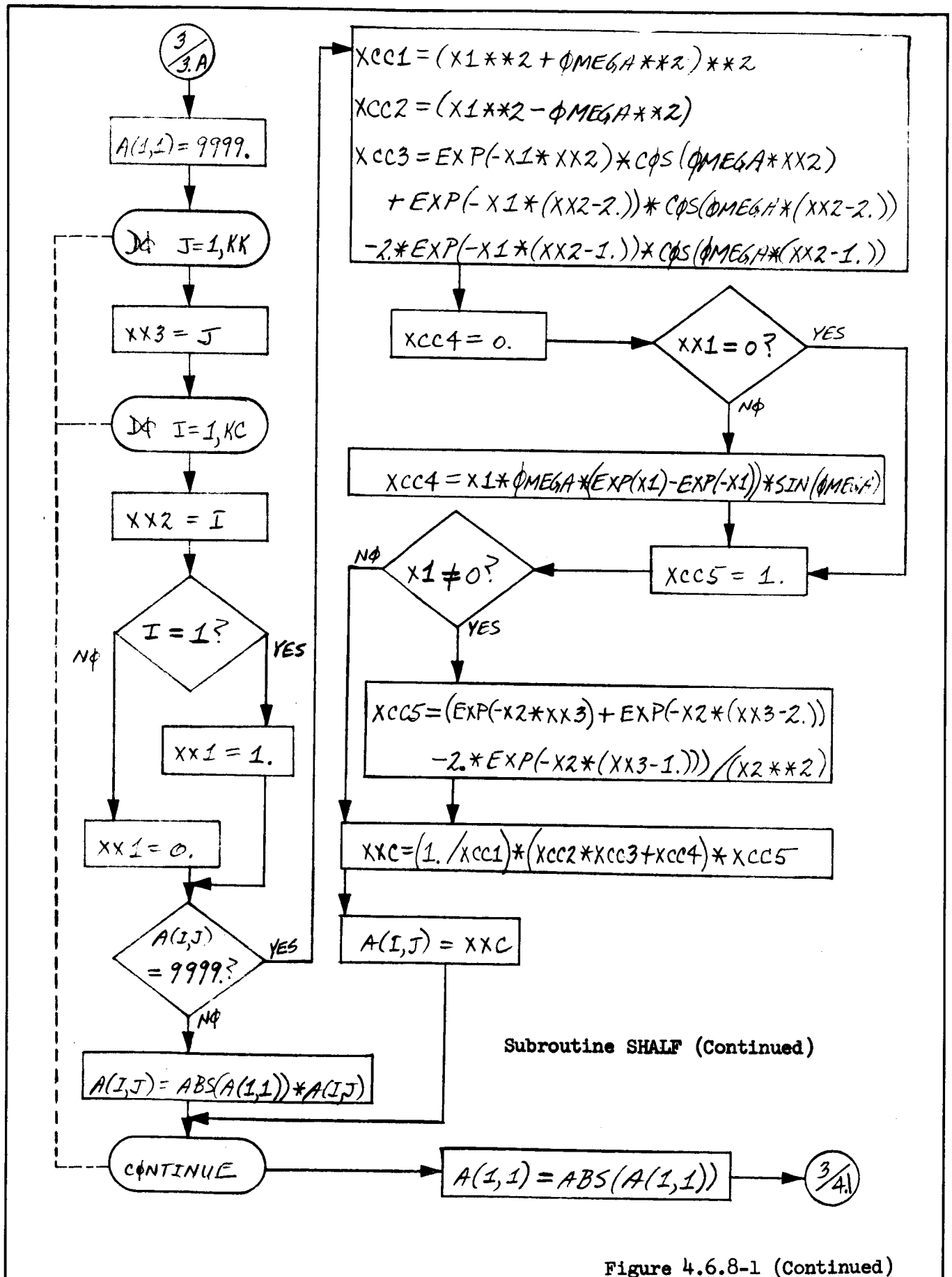


Figure 4.6.8-1 (Continued)

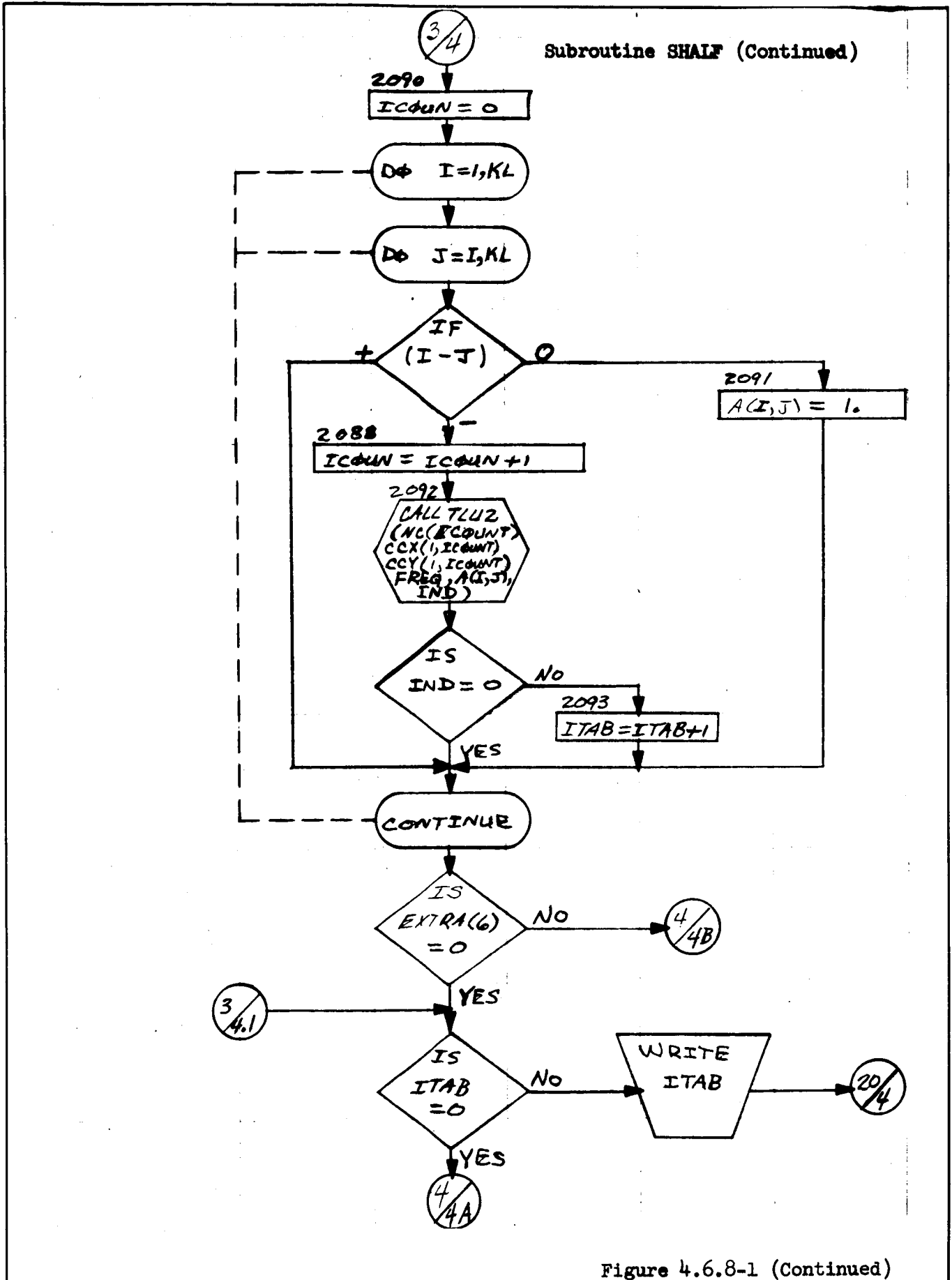


Figure 4.6.8-1 (Continued)

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Subroutine SHALF (Continued)

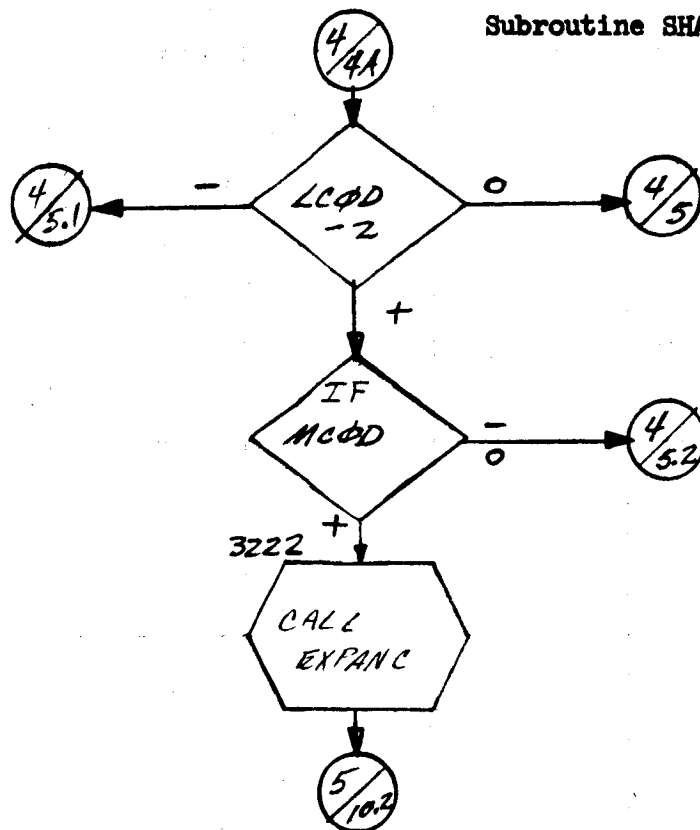


Figure 4.6.8-1 (Continued)

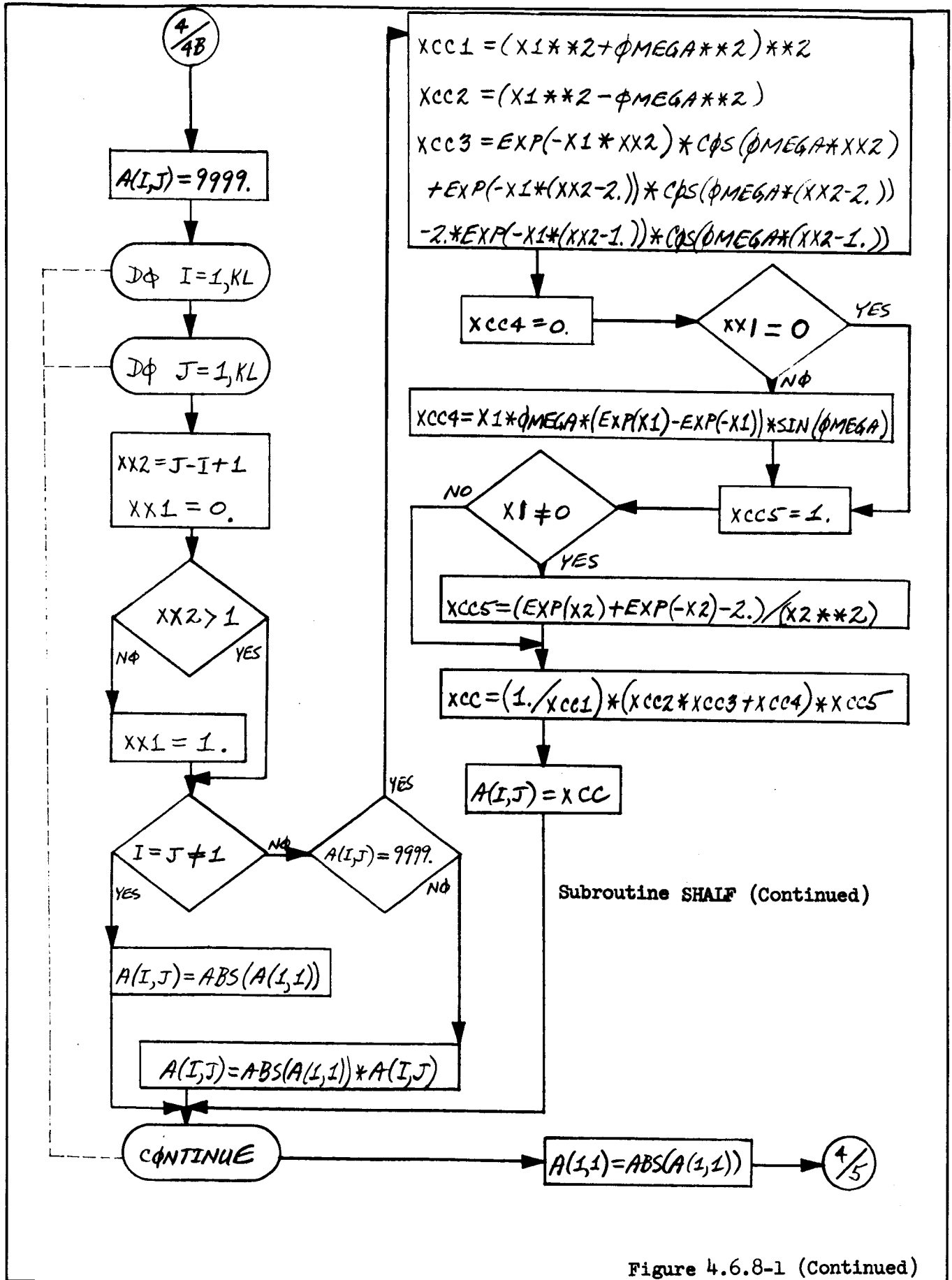


Figure 4.6.8-1 (Continued)

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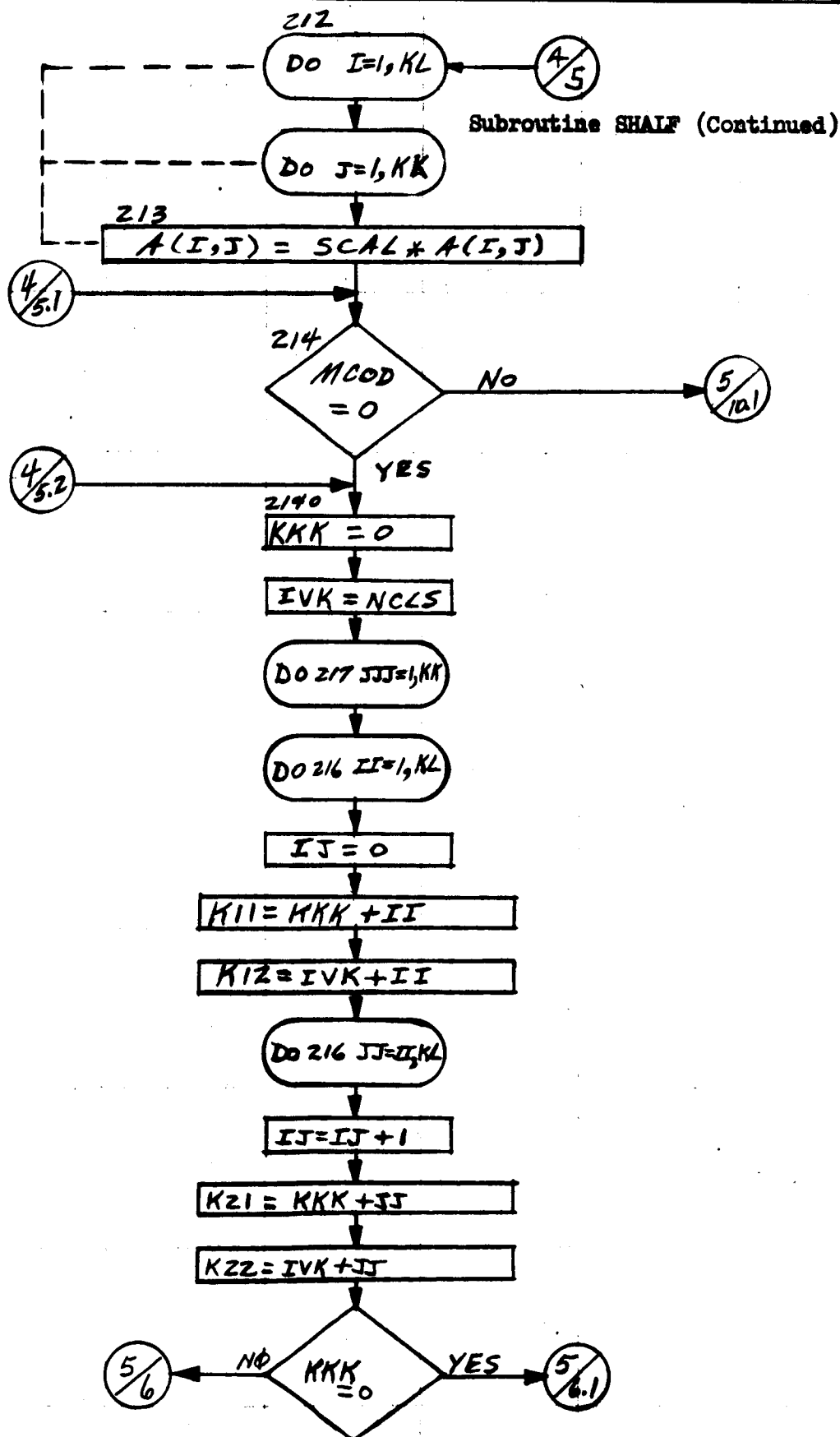


Figure 4.6.8-1 (Continued)

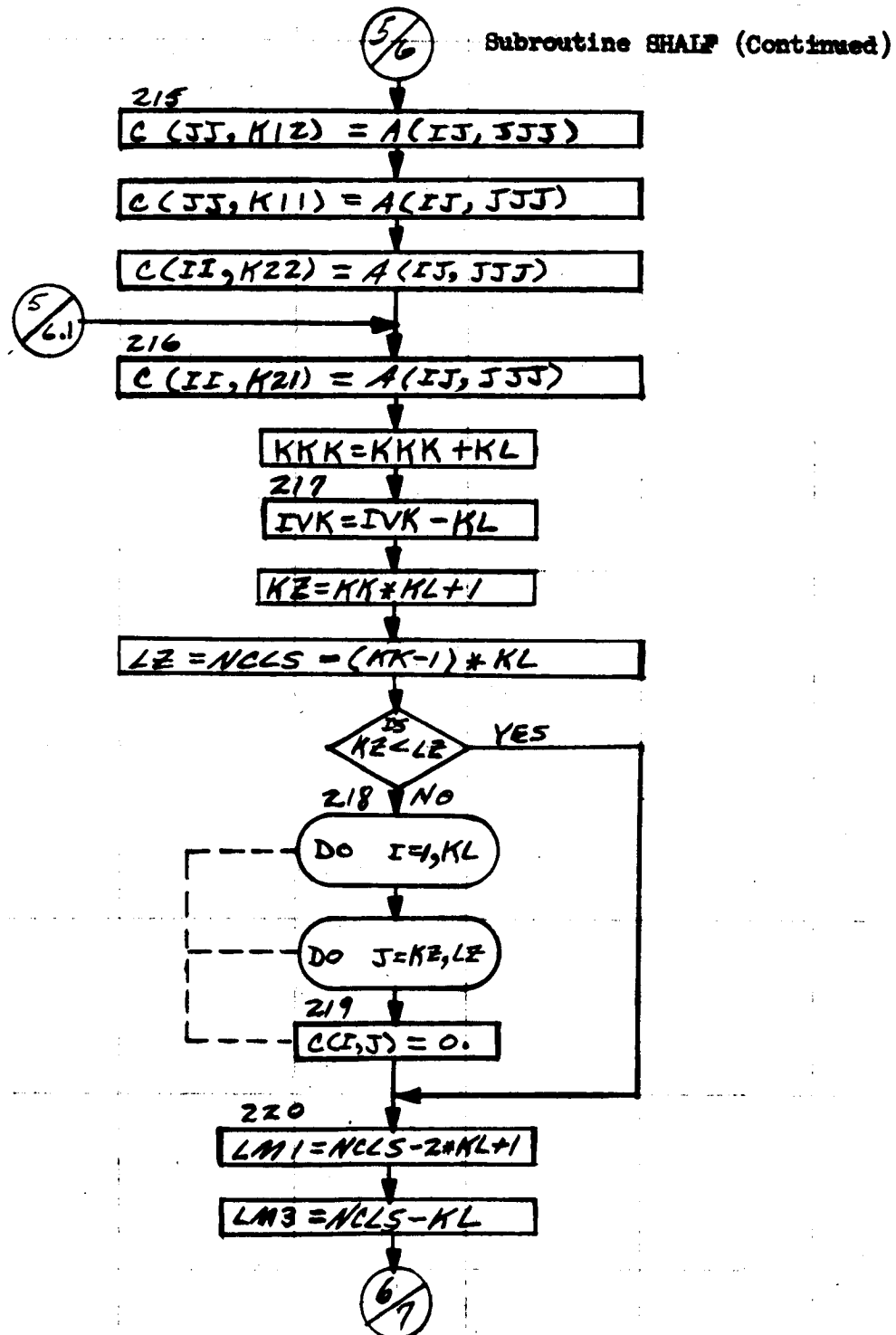


Figure 4.6.8-1 (Continued)

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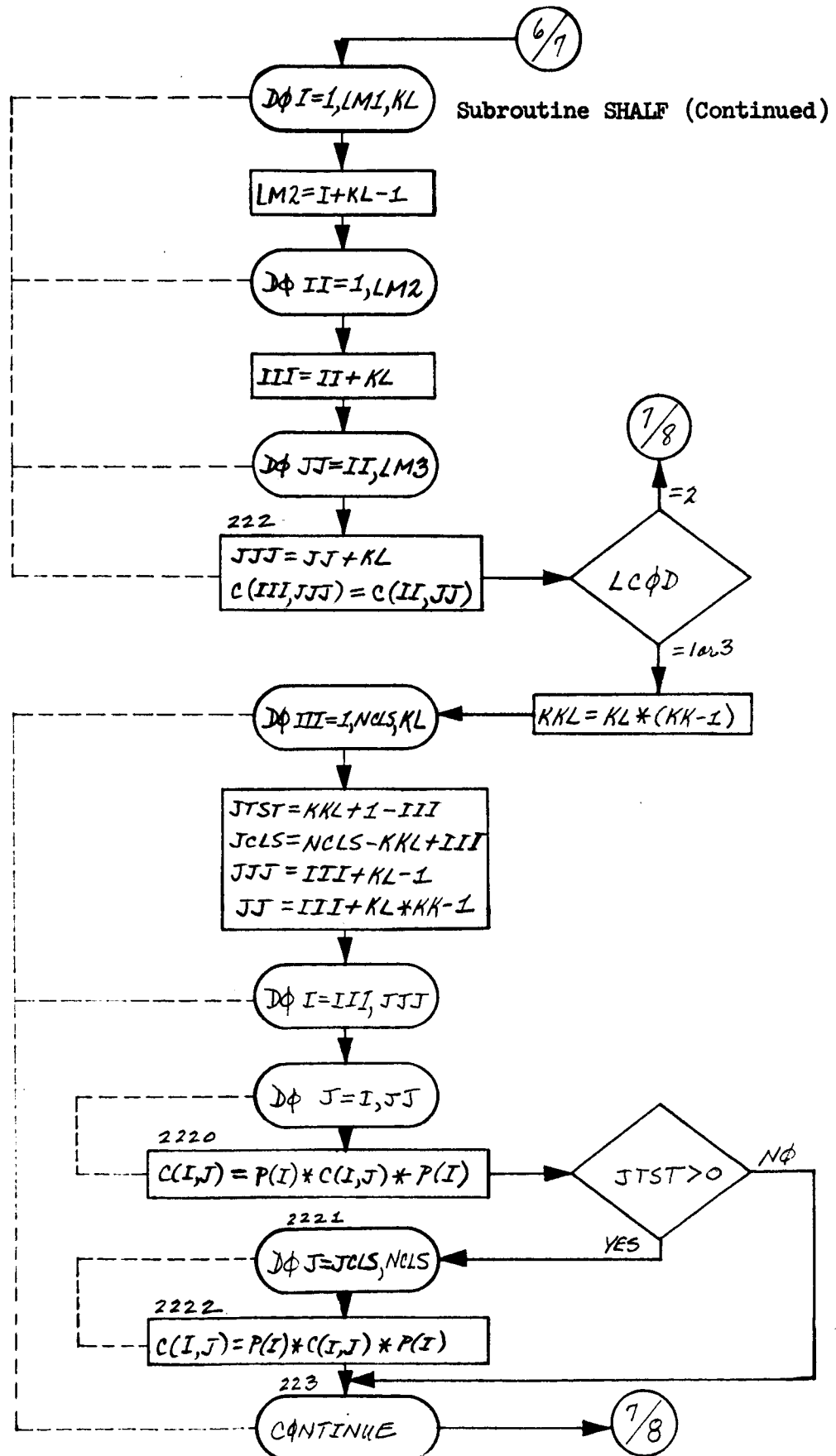


Figure 4.6.8-1 (Continued)

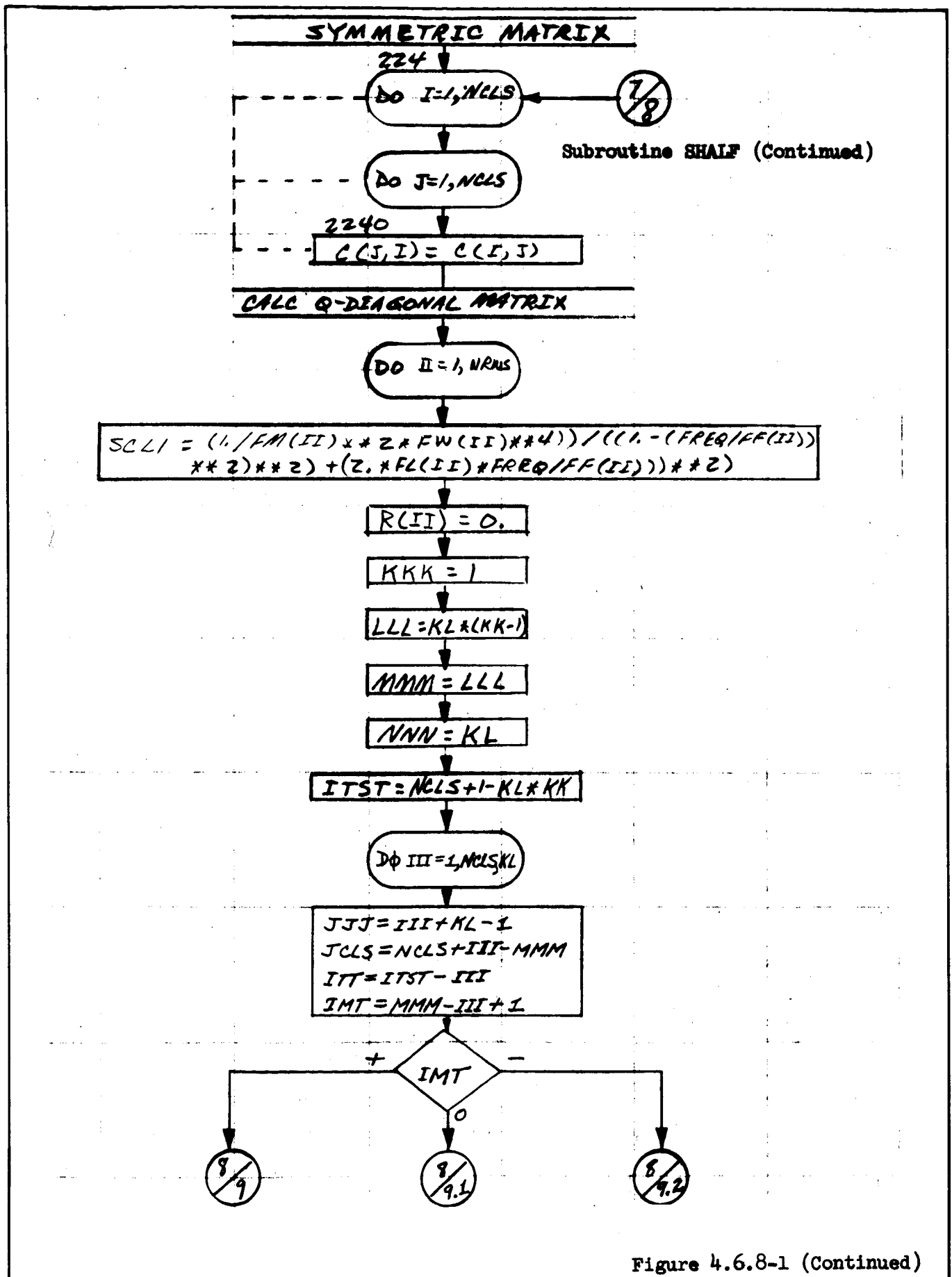


Figure 4.6.8-1 (Continued)

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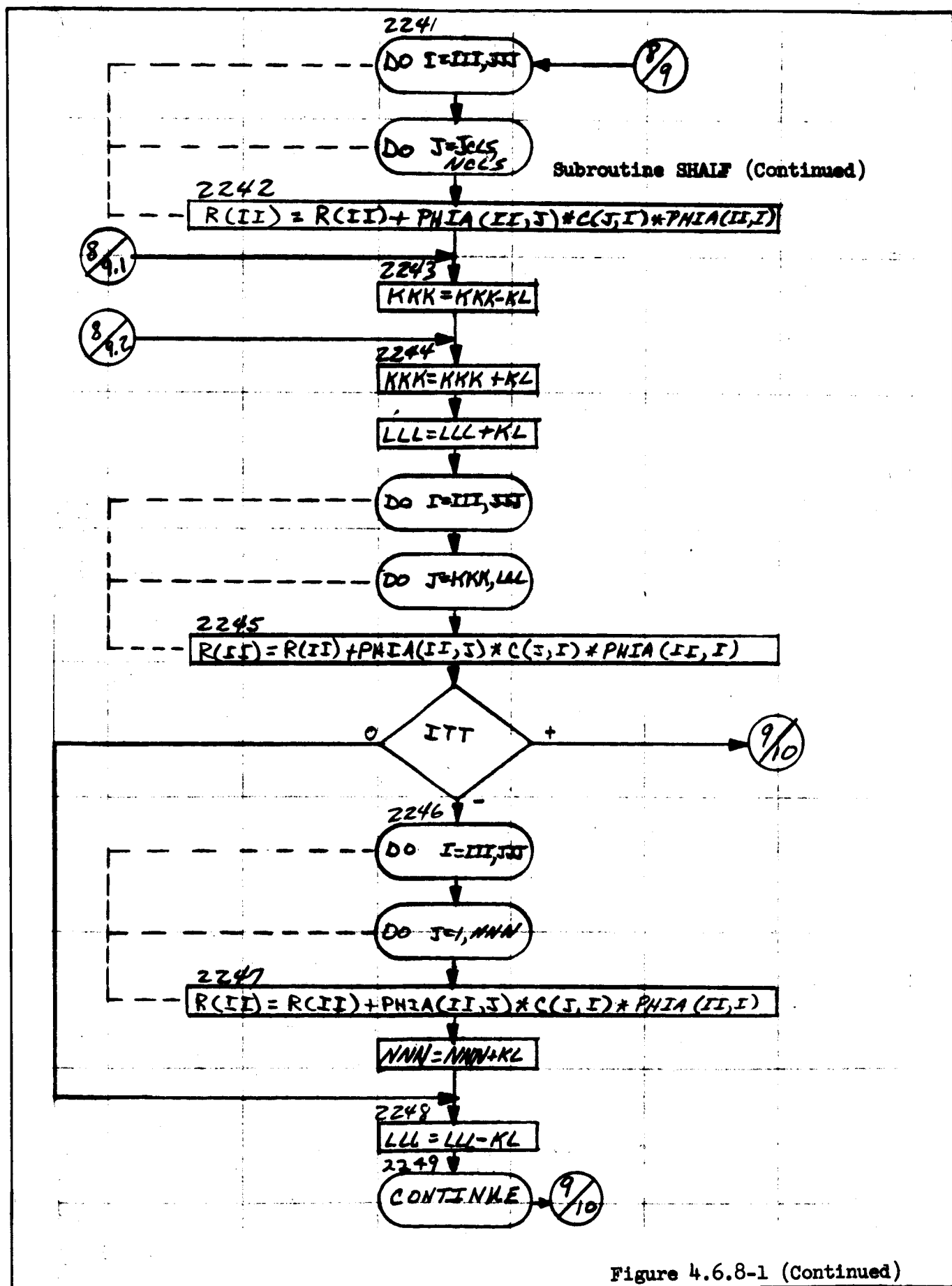


Figure 4.6.8-1 (Continued)

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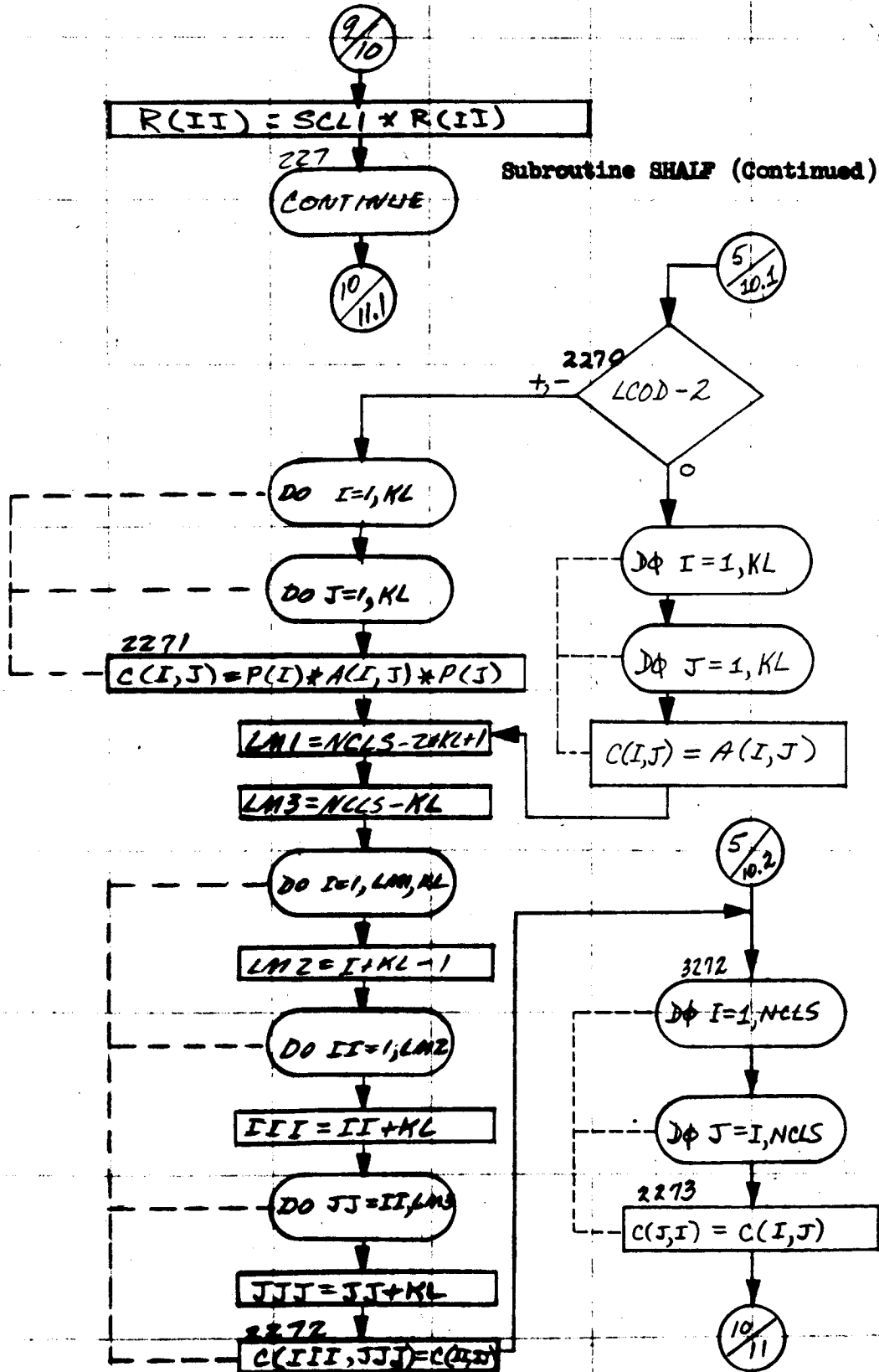


Figure 4.6.8-1 (Continued)

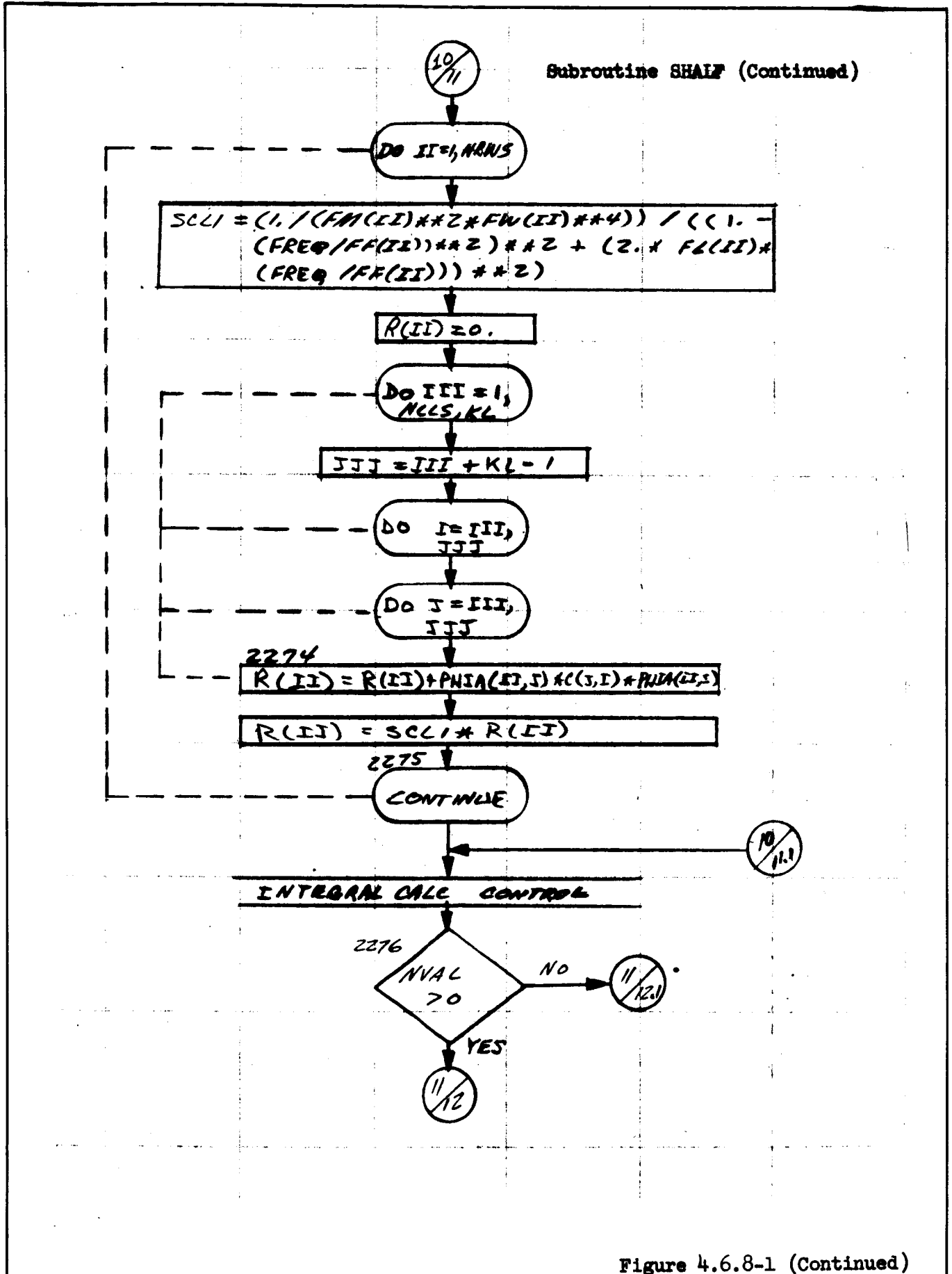


Figure 4.6.8-1 (Continued)

Subroutine SHALF (Continued)

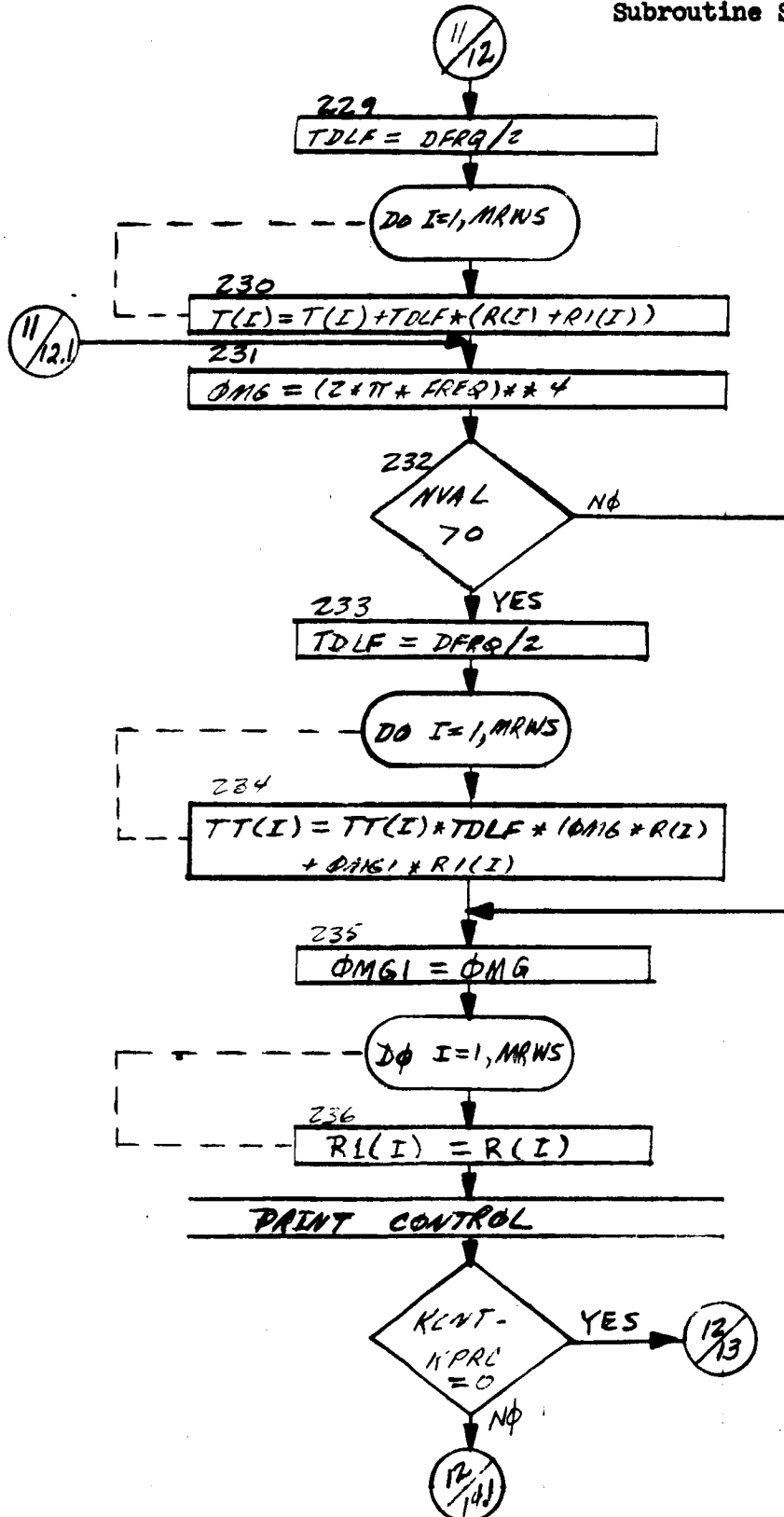


Figure 4.6.8-1 (Continued)

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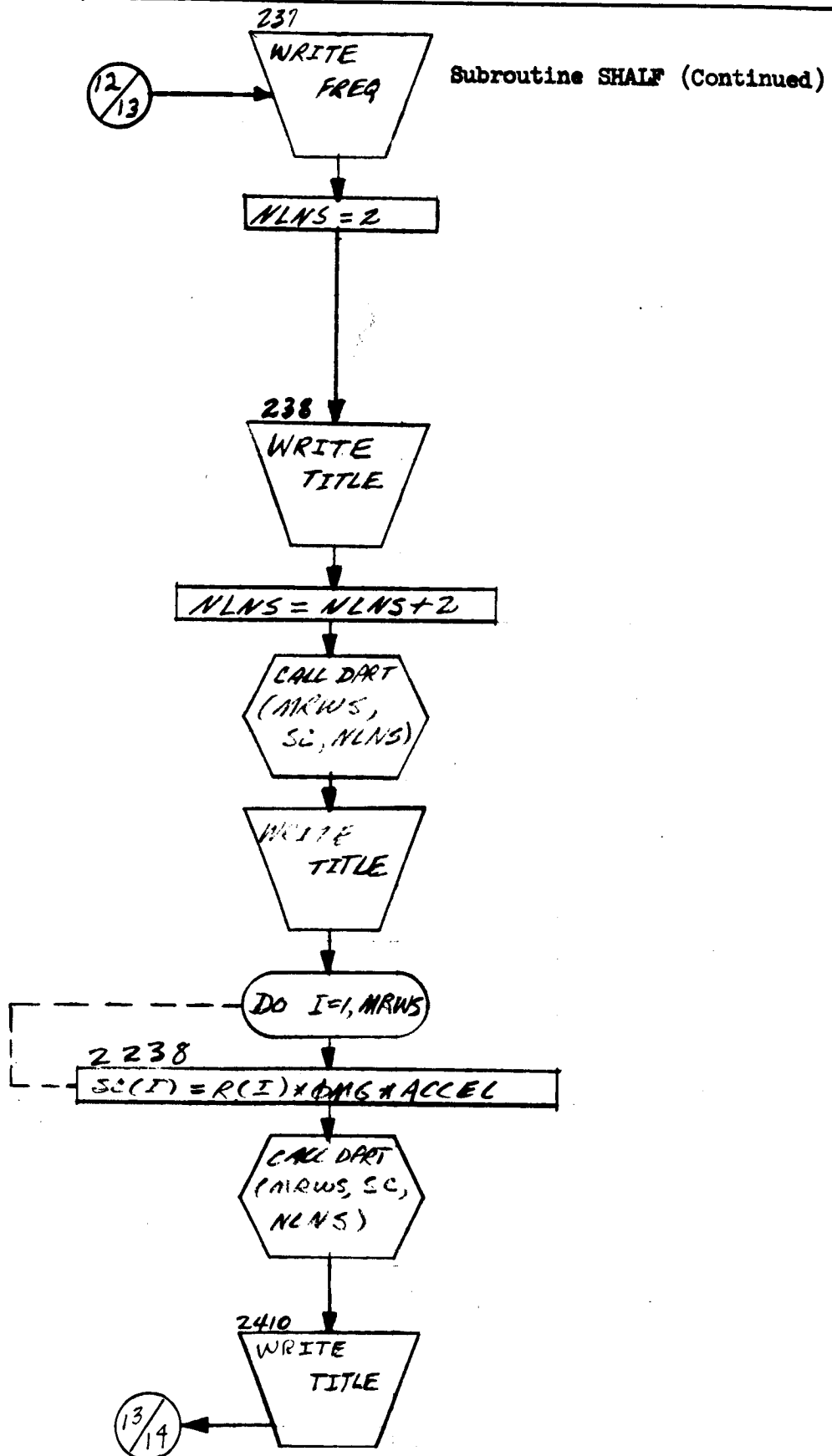


Figure 4.6.8-1 (Continued)

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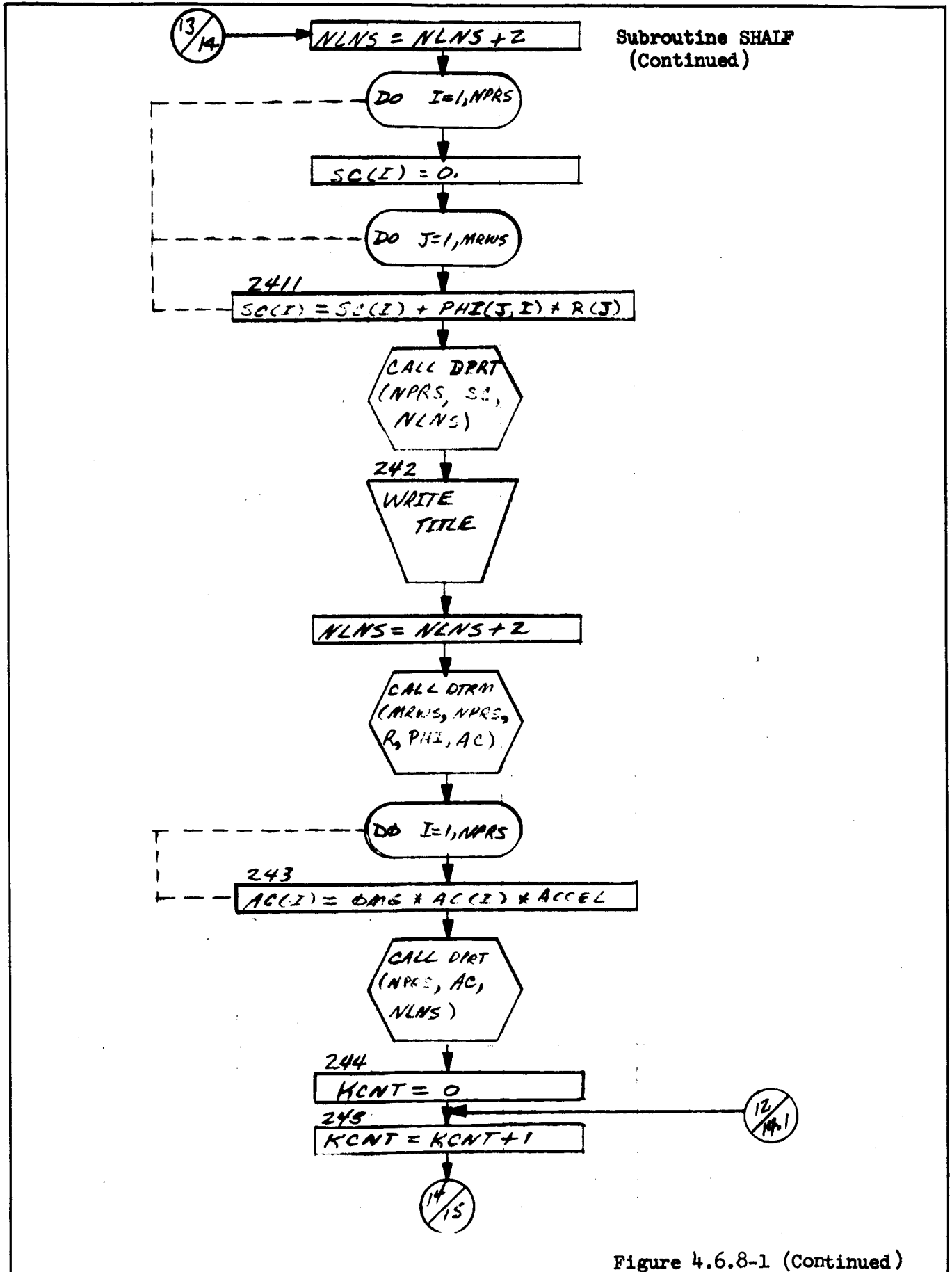


Figure 4.6.8-1 (Continued)

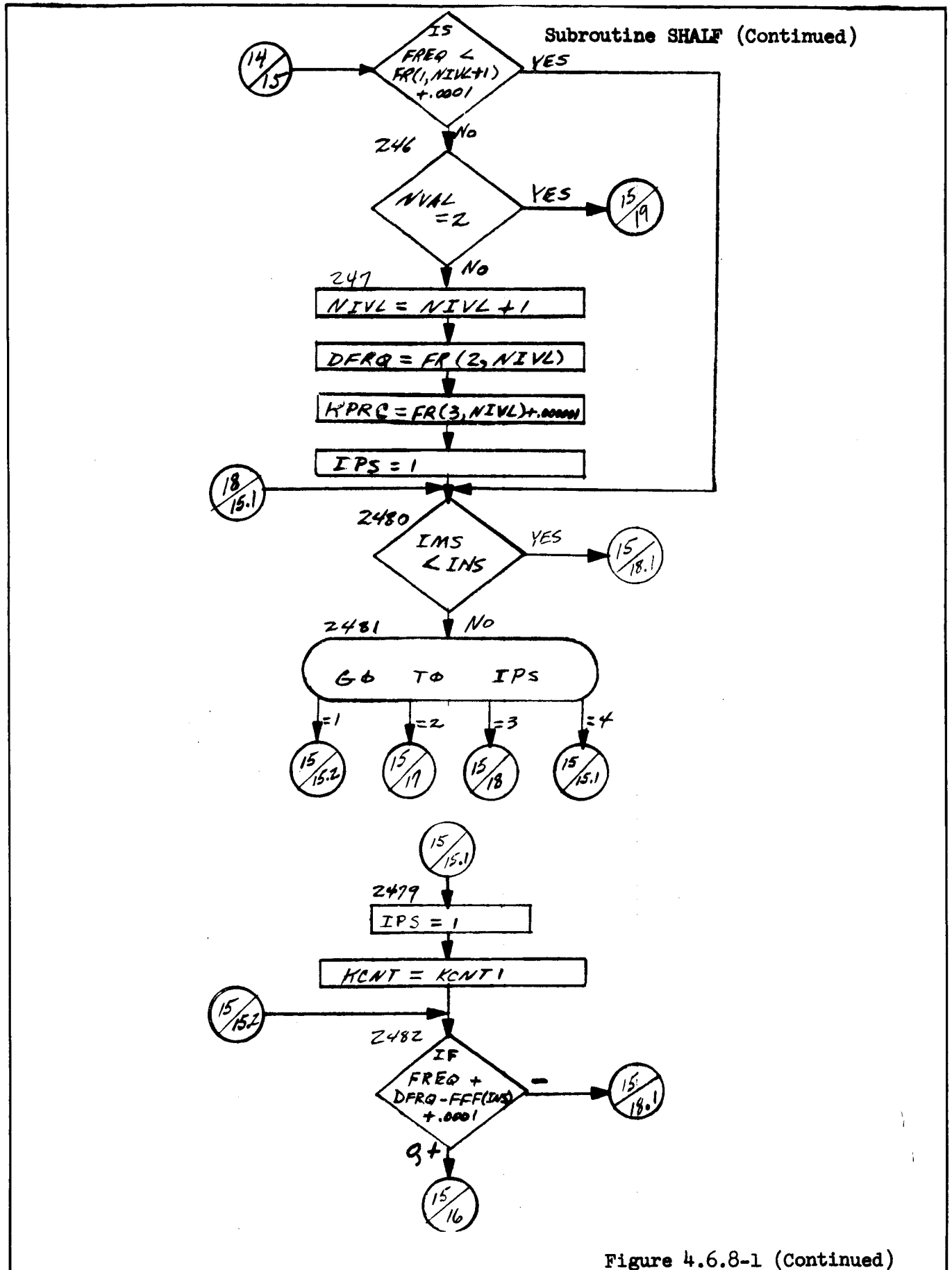


Figure 4.6.8-1 (Continued)

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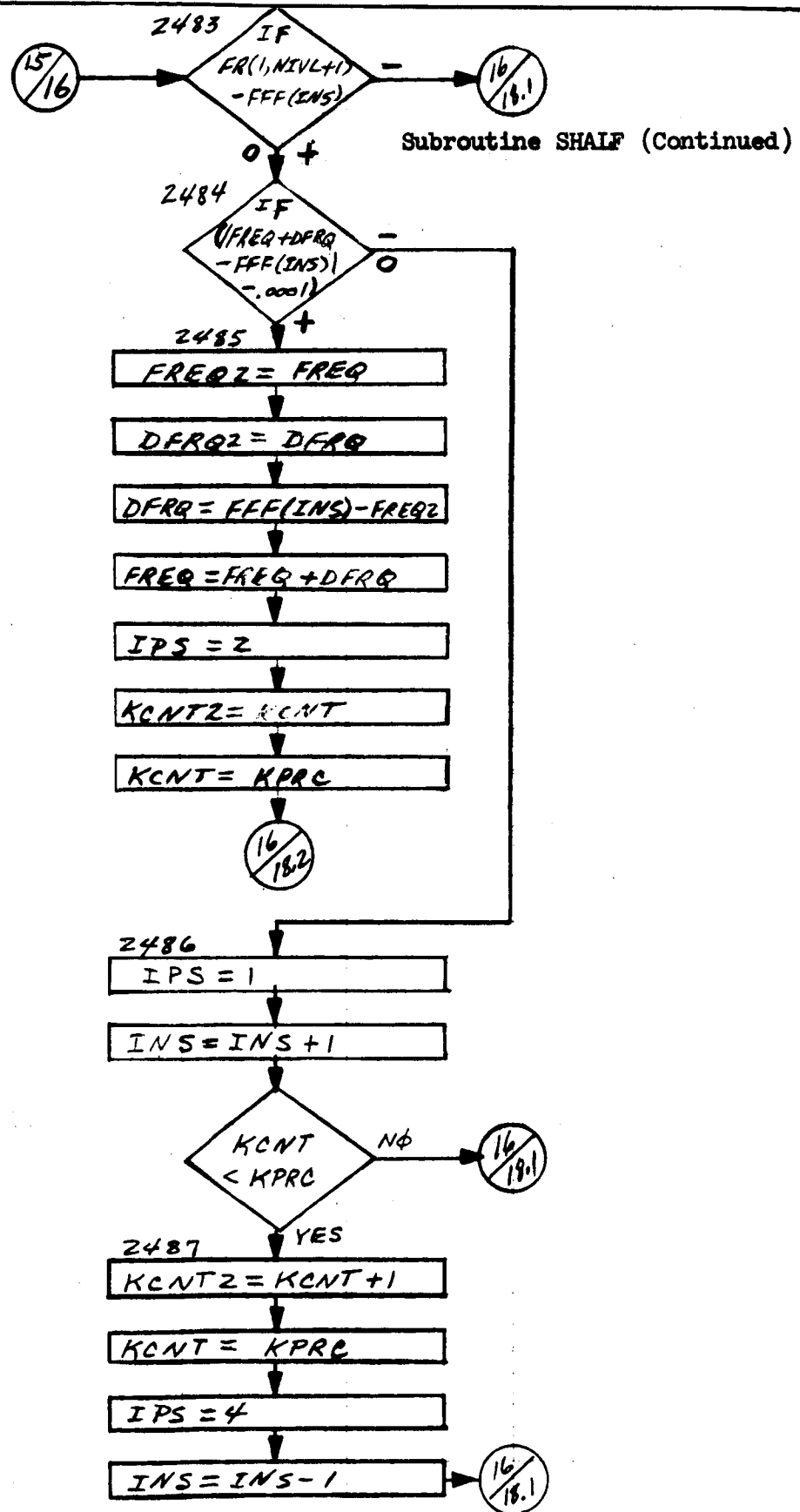


Figure 4.6.8-1 (Continued)

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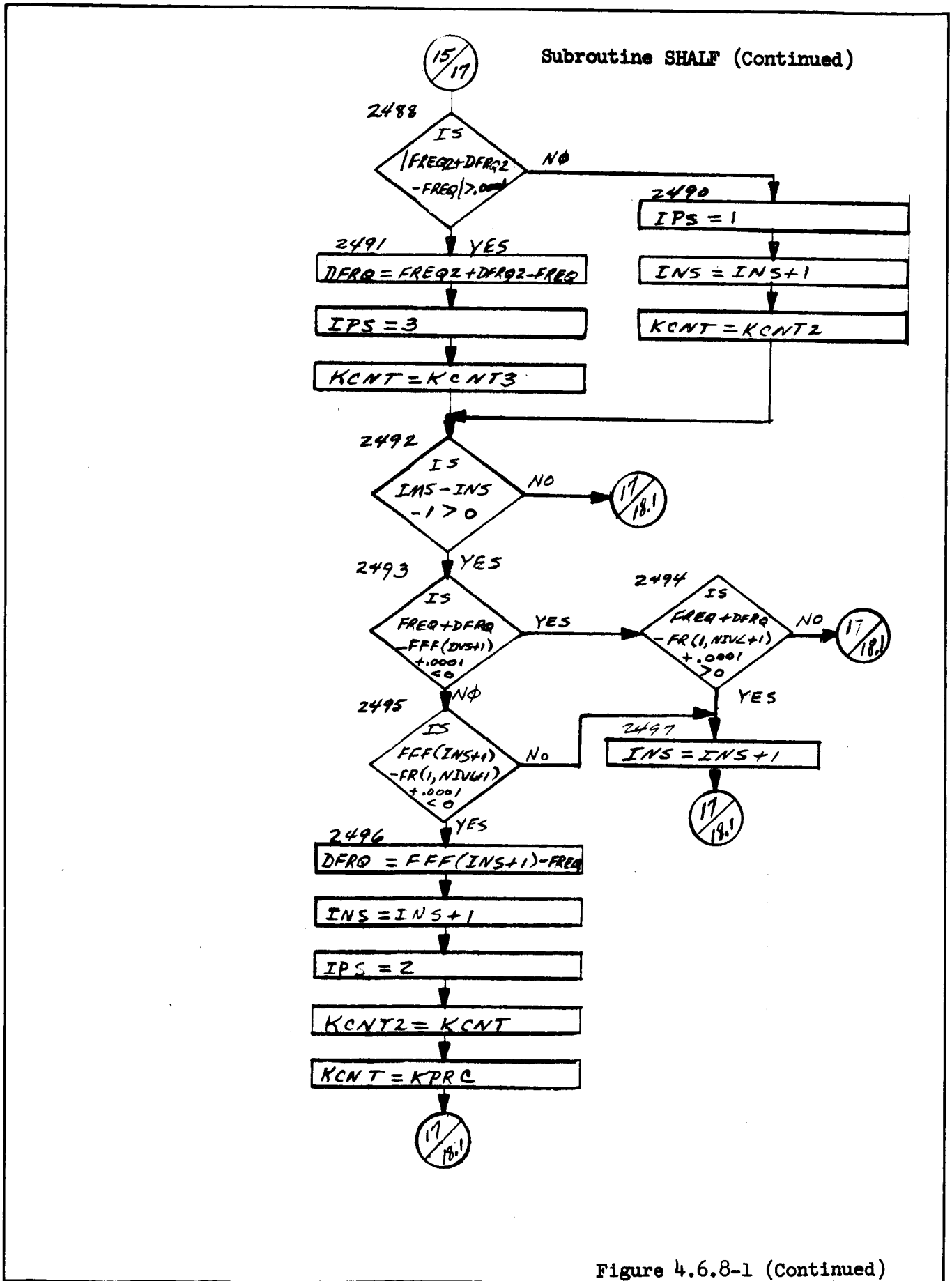


Figure 4.6.8-1 (Continued)

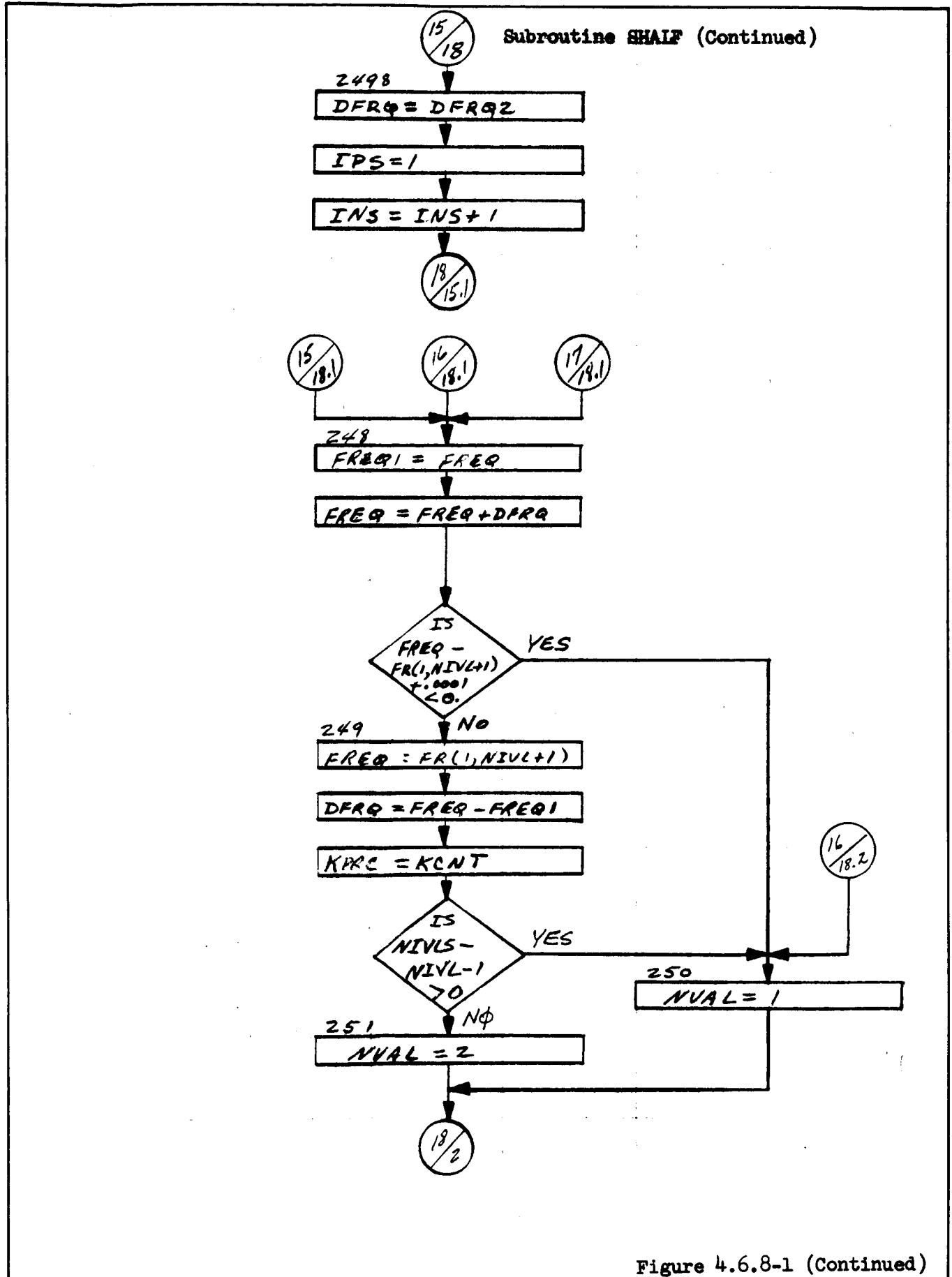


Figure 4.6.8-1 (Continued)

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Subroutine SHALF (Continued)

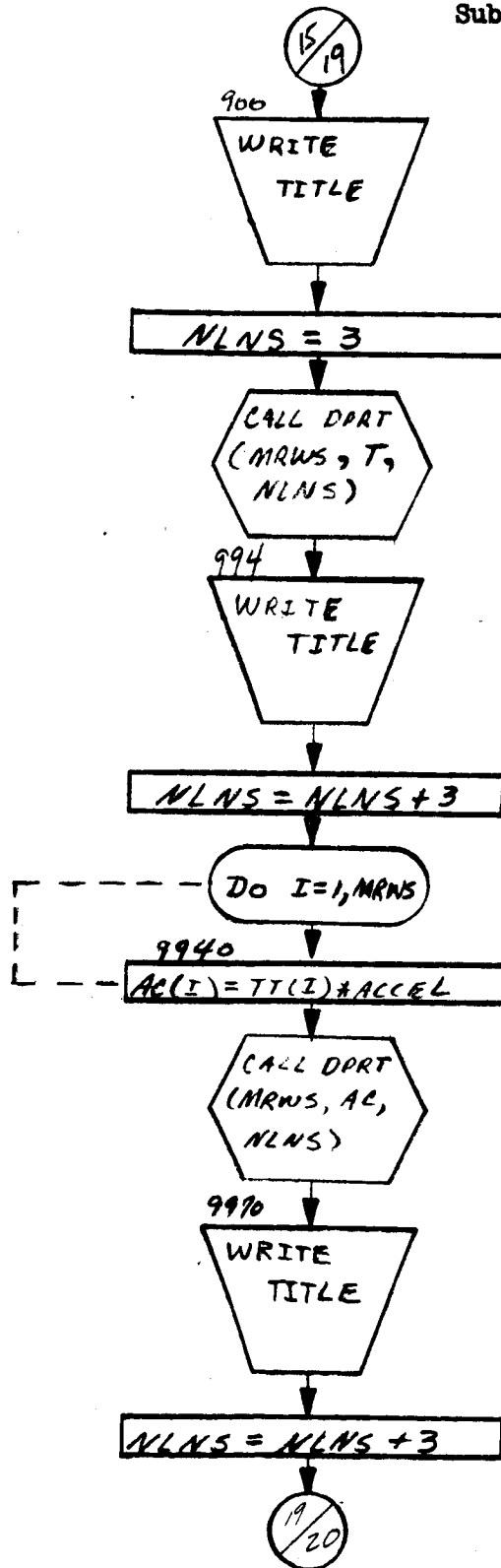


Figure 4.6.8-1 (Continued)

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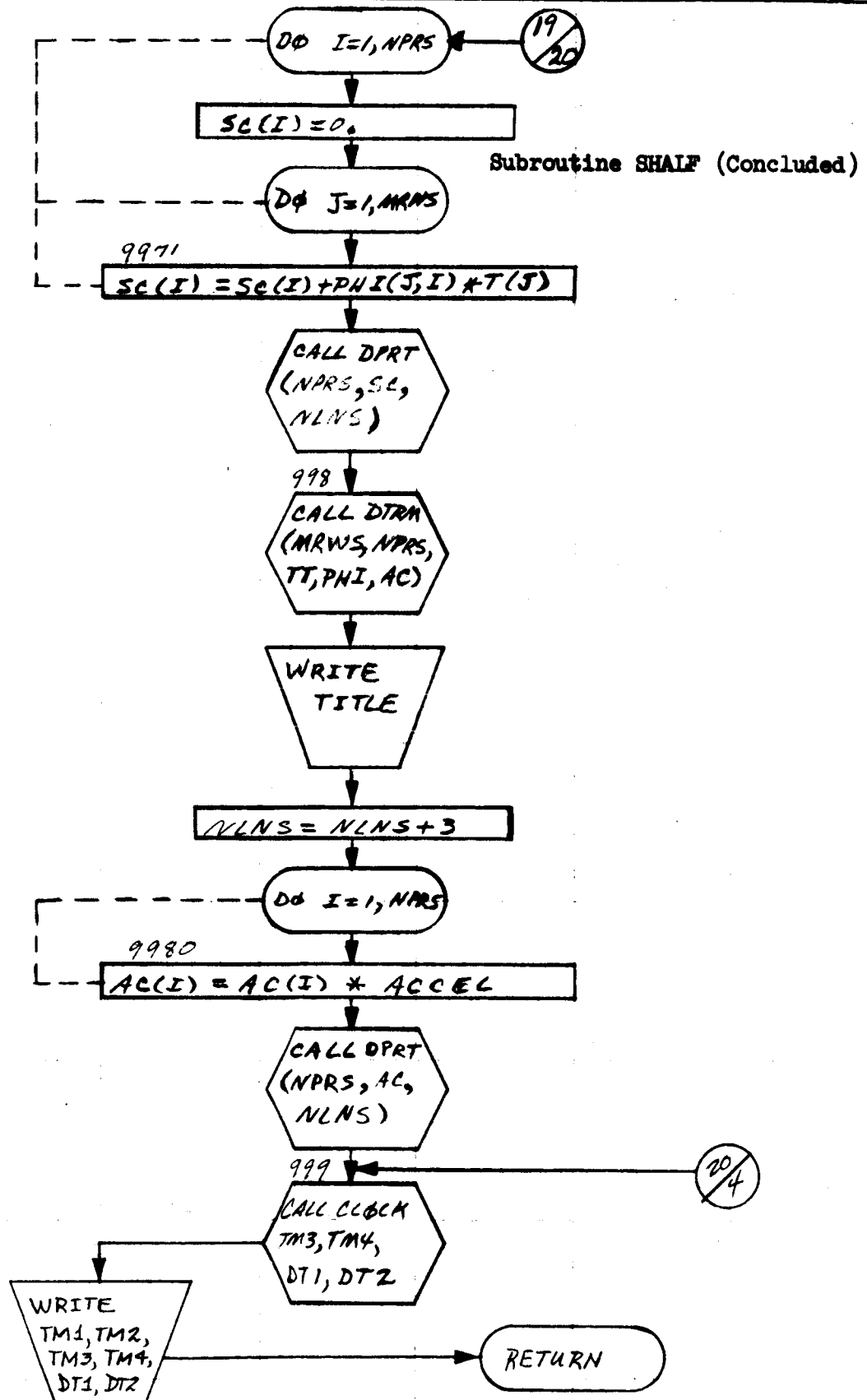


Figure 4.6.8-1 (Concluded)

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4.7 Nomenclature Table. - To adequately examine the DARC program in detail it is necessary to first be able to correlate the various parameters (inputs, outputs, and internal) with the designated symbols used in the program. This section is an attempt to show that correlation.

4.7.1 Table Description. - Table 4.7.1-1 shows a list of symbols used in the DARC program. Along with each symbol is the dimension given the parameter in the program and a comment stating its usage or its description.

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NOMENCLATURE TABLE

0010	INPUT MATRIX OF NORMALIZED CO-SPECTRA
0020	ACCELERATION SPECTRA
0030	SCALE FACTOR ON THE ACCELERATION RESPONSE MATRIX
0040	INTERMEDIATE STORAGE
0050	MATRIX OF CO-SPECTRA OF PRESSURES
0060	INTERMEDIATE STORAGE
0070	COS OF COMPLEMENT OF CONE ANGLE
0080	INTERMEDIATE STORAGE
0090	INTERMEDIATE STORAGE
0100	INDEPENDENT VARIABLE, FREQUENCY (CO-SPECTRUM TABLE)
0110	DEPENDENT VARIABLE (CO-SPECTRUM TABLE)
0120	LENGTH OF FRUSTUM / PI
0130	COS OF COMPLEMENT OF CONE ANGLE SQUARED
0140	DELTA FREQUENCY
0150	DELTA FREQUENCY TEMPORARY STORAGE
0160	LENGTH OF FRUSTUM SEGMENT
0170	NUMBER OF DEGREES IN EACH SEGMENT
0180	DATE TAKEN FROM CLOCK
0190	DATE TAKEN FROM CLOCK
0200	CO-SPECTRA EQUATION PARAMETERS
0210	1-DECAY RATE IN LONGITUDINAL DIRECTION
0220	2-DECAY RATE IN CIRCUMFERENTIAL DIRECTION
0230	3-AVERAGE SIZE OF AREA (LONG.)
0240	4-AVERAGE SIZE OF AREA (CIRC.)
0250	5-CONVECTION VELOCITY
0260	6-SWITCH TO DETERMINE METHOD TO CALC. CO-SPECTRA
0270	7-10 NOT USED
0280	SCALE FACTOR ON INDEPENDENT VARIABLE (ALL TABLES)
0290	SCALE FACTOR ON PSD VALUES ONLY
0300	COMPLEMENT OF CONE ANGLE DEGREES
0310	NORMAL MODE FREQUENCY
0320	NORMAL MODE FREQUENCY TEMPORARY STORAGE
0330	INTERMEDIATE STORAGE
0340	NORMAL MODE DAMPING COEFFICIENT
0350	INTERMEDIATE STORAGE
0360	LENGTH OF FRUSTUM / (LONG. HALF-WAVE*PI)**3
0370	LENGTH OF FRUSTUM
0380	RADIAL MODE COEFFICIENT FOR (N,M)

7,7

25

A

AC

ACCEL

BETA

112,112

C

CA

CALP

CB

CBTA

20,21

CCX

20,21

CCY

CFLMT

CSQ

DFRQ

DFRQ2

DS

DIHT

DT1

DT2

10

EXTRA

FAC1

FAC2

FALP

25

FF

25

FFF

FFF1

25

FL

FLMC

FLMT

FLNG

FLTA

Table 4.7.1-1

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NOMENCLATURE TABLE (Continued)

FLIM	25	NUMBER OF LONGITUDINAL HALF WAVES	0410
FLTN		NUMBER OF CIRCUMFERENTIAL WAVES	0420
FLIS		NS FLOATED	0430
FLIT		NT FLOATED	0440
FM	25	NORMAL MODE GENERALIZED MASS	0450
FMD	3.25	NUMBER OF LONGITUDINAL HALF WAVES	0460
FR	3.50	FREQUENCY DATA	0470
FREQ		FREQUENCY	0480
FREQ01		FREQUENCY TEMPORARY STORAGE	0490
FREQ02		FREQUENCY TEMPORARY STORAGE	0500
FW	25	NORMAL MODE FREQUENCY * 6.28318531	0510
FX1		INTERMEDIATE STORAGE	0520
FX2		INTERMEDIATE STORAGE	0530
F11		INTERMEDIATE STORAGE	0540
F12		INTERMEDIATE STORAGE	0550
F21		INTERMEDIATE STORAGE	0560
F22		INTERMEDIATE STORAGE	0570
HD	10	CASE ID	0580
I		INDEX REGISTER	0590
ICOD		PHIA TRANSPOSE CODE	0600
ICOUN		COUNTER	0610
II		COUNTER	0620
III		COUNTER	0630
IJ		COUNTER	0640
IK		COUNTER	0650
IMS		COUNTER	0660
IMT		COUNTER	0670
IND		ERROR INDICATOR	0680
INS		COUNTER	0690
IPS		COUNTER	0700
IR		COUNTER	0710
IRWS		NUMBER OF MODES	0720
ITAB		TOTAL NO. ERRORS	0730
ITST		COUNTER	0740
ITT		COUNTER	0750
IVK		COUNTER	0760
J		INDEX REGISTER	0770
JCCD		JCOD+1	0780
JCLS		COUNTER	0790
JCOD		PHI*#2 TRANSPOSE CODE	0800

Table 4.7.1-1 (Continued)

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NOMENCLATURE TABLE (Continued)

0810	COUNTER	JJ
0820	COUNTER	JJJ
0830	NUMBER OF MODES	JRWS
0840	COUNTER	J1ST
0850	INDEX REGISTER	K
0860	NUMBER OF NON-ZERO LONGITUDINAL AREAS	KC
0870	COUNTER	KCNT
0880	COUNTER	KCNT1
0890	COUNTER	KCNT2
0900	NUMBER OF RADIAL AREAS	KK
0910	COUNTER	KKK
0920	COUNTER	KKL
0930	NUMBER OF LONGITUDINAL AREAS	KL
0940	COUNTER	KLNS
0950	COUNTER	KPRC
0960	NUMBER OF MODES	KPRS
0970	NUMBER OF MODES	KRWS
0980	COUNTER	KZ
0990	COUNTER	K11
1000	COUNTER	K12
1010	COUNTER	K21
1020	COUNTER	K22
1030	COUNTER	L
1040	COUNTER	LL
1050	PSD INPUT CODE	LCOD
1060	COUNTER	LLL
1070	COUNTER	LM1
1080	COUNTER	LM2
1090	COUNTER	LM3
1100	COUNTER	LZ
1110	CO-CORRELATION EXPANSION CODE	MCOD
1120	COUNTER	MMM
1130	NUMBER OF MODES	MRWS
1140	NUMBER OF MODES	N
1150	NUMBER OF POINTS IN CO-SPECTRUM TABLE	NC
1160	CASE NUMBER	NCASE
1170	NUMBER OF SURFACE AREAS	NCLS
1180	NUMBER OF COEFFICIENTS USED XC(1,I)	NCN
1190	DUMMY VARIABLE	NDUM
1200	COUNTER	NIVL

Table 4.7.1-1 (Continued)

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NOMENCLATURE TABLE (Continued)

1210			
1220			
1230			
1240			
1250			
1260			
1270			
1280			
1290			
1300			
1310			
1320			
1330			
1340			
1350			
1360			
1370			
1380			
1390			
1400			
1410			
1420			
1430			
1440			
1450			
1460			
1470			
1480			
1490			
1500			
1510			
1520			
1530			
1540			
1550			
1560			
1570			
1580			
1590			
1600			
NIVLS			
NLN	25	NUMBER OF FREQUENCY VALUES	
NLNS		NUMBER OF COEFFICIENTS USED XA(1,I)	
NNN		COUNTER	
NNS		INTERMEDIATE STORAGE	
NNS1		INTERMEDIATE STORAGE	
NNTH		INTERMEDIATE STORAGE	
NNTH1		INTERMEDIATE STORAGE	
NP	112	NUMBER OF POINTS IN PSD TABLE	
NPRQ		INTERMEDIATE PRINTOUT CODE	
NPRS		NUMBER OF ACCELERATION RESPONSE POINTS	
NRWS		NUMBER OF MODES	
NS		NUMBER OF LONGITUDINAL VALUES	
NSTS		NUMBER OF LONGITUDINAL VALUES	
NT		NUMBER OF RADIAL VALUES	
NVAL		COUNTER	
NXCO		COUNTER	
N1		COUNTER	
OMG		INTERMEDIATE STORAGE	
OMG1		INTERMEDIATE STORAGE	
OMEGA		INTERMEDIATE STORAGE	
P	112	DIAGONAL MATRIX OF SORTF(PSD) AT FREQ IN QUESTION	
PHA		HOLLERITH REPRESENTATION OF PHIA	
PHI	25,20	MODAL DEFLECTION SQUARED MATRIX	
PHIA	25,112	TRANSFORMATION MATRIX TO CONVERT THE PRESSURE SPECTRUMS TO GENERALIZED FORCE SPECTRUMS	
PHI2		HOLLERITH REPRESENTATION OF PHI2	
PX	20,112	INDEPENDENT VARIABLE, FREQUENCY (PSD TABLE)	
PY	20,112	DEPENDENT VARIABLE (PSD TABLE)	
R	25	NORMAL COORDINATE SPECTRA	
R1	25	NORMAL COORDINATE SPECTRA PREVIOUS FREQUENCY	
SALP	25	SIN OF COMPLEMENT OF CONE ANGLE	
SC		DEFLECTION SPECTRA	
SCAL		SCALAR USED TO MULT. CO-SPECTRUM SUR MATRIX	
SC11		INTERMEDIATE STORAGE	
STOT		DISTANCE FROM APEX AT THE POINT IN QUESTION	
SVAL	20	DISTANCE BETWEEN APEX AND GIVEN VALUE	
S1		DISTANCE BETWEEN THE APEX AND INITIAL VALUE (X1)	
S1C		INTERMEDIATE STORAGE	
S1T	112	INTERMEDIATE STORAGE	

Table 4.7.1-1 (Continued)

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1610			
1620			
1630			
1640			
1650			
1660			
1670			
1680			
1690			
1700			
1710			
1720			
1730			
1740			
1750			
1760			
1770			
1780			
1790			
1800			
1810			
1820			
1830			
1840			
1850			
1860			
1870			
1880			
1890			
1900			
1910			
1920			
1930			
1940			
1950			
1960			
1970			
1980			
1990			
2000			
S2			DISTANCE BETWEEN THE APEX AND FINAL VALUE (XF)
S2C			INTERMEDIATE STORAGE
S2T	112		INTERMEDIATE STORAGE
T	25		INTERMEDIATE STORAGE
TDLF			DELTA FREQUENCY DIVIDED BY TWO
TH1T			THETA IN DEGREES
TH1			CONSTANT = 0.
TH2			CONSTANT = 360.
TINT	112		INTEGRAL OF THETA FUNCTION
TM1			TIME TAKEN FROM CLOCK
TM2			TIME TAKEN FROM CLOCK
TM3			TIME TAKEN FROM CLOCK
TM4			TIME TAKEN FROM CLOCK
TT	25		INTERMEDIATE STORAGE
X			COMPLEMENT OF CONE ANGLE RADIAN
X1			EXTRA(1)*EXTRA(3)
X2			EXTRA(2)*EXTRA(4)
XA	25,31		COEFFICIENTS USED IN FOURIER SERIES CALCULATION OF PHIA
XAM			INTERMEDIATE STORAGE
XANGLE			INTERMEDIATE STORAGE
XB	25,30		COEFFICIENTS USED IN FOURIER SERIES CALCULATION OF PHIA
XBM			INTERMEDIATE STORAGE
XC	25,31		COEFFICIENTS USED IN FOURIER SERIES CALCULATION OF PHIA
XCC1			SEGMENT OF EMPIRICAL CO-SPECTRA EQN
XCC2			SEGMENT OF EMPIRICAL CO-SPECTRA EQN
XCC3			SEGMENT OF EMPIRICAL CO-SPECTRA EQN
XCC4			SEGMENT OF EMPIRICAL CO-SPECTRA EQN
XCC5			SEGMENT OF EMPIRICAL CO-SPECTRA EQN
XD	25,30		COEFFICIENTS USED IN FOURIER SERIES CALCULATION OF PHIA
XDELS			LENGTH EACH AREA ALONG THE ADAPTER
XDELTA			COORDINATE TO WHICH FOURIER SERIES IS REFERENCED
XDTH			DELTA THETA (THETA = ANGLE OF ADAPTER)
XKM	25		INTERMEDIATE STORAGE USED IN FOURIER SERIES
XKN	112		INTERMEDIATE STORAGE USED IN FOURIER SERIES
XL			LENGTH OF ADAPTER
XL1			CO-SPECTRA EQN LONGITUDINAL DIRECTION
XLM	25		INTERMEDIATE STORAGE USED IN FOURIER SERIES
XLN	112		INTERMEDIATE STORAGE USED IN FOURIER SERIES
XM	25,30		COEFFICIENTS USED IN FOURIER SERIES CALCULATION OF PHIA
XN	25,30		COEFFICIENTS USED IN FOURIER SERIES CALCULATION OF PHIA

Table 4.7.1-1 (Continued)

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NOMENCLATURE TABLE (Concluded)

2010
2020
2030
2040
2050
2060
2070
2080
2090
2100
2110
2120
2130
2140
2150

XNS	FLOATED NS
XNT	FLOATED NT
XQ	INTERMEDIATE STORAGE
XSU	USED FOR INTERMEDIATE STORAGE
XSUM	USED FOR INTERMEDIATE STORAGE
XSUM1	INTERMEDIATE STORAGE USED IN FOURIER SERIES
XSUM2	INTERMEDIATE STORAGE USED IN FOURIER SERIES
XT1	CO-SPECTRA EQN CIRCUMFERENTIAL DIRECTION
XTHFTA	ANGLE OF ADAPTER
XTIME	COUNTER
XX1	CO-SPECTRA EQN, DELTA K
XX2	CO-SPECTRA EQN, KL (OR KC)
XX3	CO-SPECTRA EQN, KK
XXC	CO-SPECTRA EQN VALUE
XXR	RADIUS OF ADAPTER

Table 4.7.1-1 (Concluded)

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4.8 Program Listing. - The DARC program listing as it exists in its symbolic form is contained in this section. It should be noted that the library routines (i.e. mathematical routines) are not contained in the listing or the actual program cards. It is assumed that these basic routines will be contained in the system wherever this program is compiled. The listing shows the make-up of the deck as it would be for compilation and execution, with the exception of an \$ID card to precede the deck and data following the deck. The data setup is described in section 4.4.2.

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```

$EXECUTE      IBJOB
$IBJOB        MAP
$IBFTC T650   REF
CT650
1             CALL PART1
              CALL FHALL
              CALL SHALL
              GO TO 1
              END
$IBFTC EXPANC REF
CEXPANSION SUBROUTINE FOR STRAIN RESPONSE CALCULATION
C
C
C
C
              DEFLECTION ACCELERATION RESPONSE CALCULATION
              T6500000
              T6500010
              T6500020
              T6500030
              T6500040
              T6500050
              T6500060
              T6500070
              T6500080
              T6500090
              T6500100
              T6500110
              T6500120
              T6500130
              T6500140
              T6500150
              T6500160
              T6500170
              T6500180
              T6500190
              T6500200
              T6500210
              T6500220
              T6500230
              T6500240
              T6500250
              T6500260
              T6500270
              T6500280
              T6500290
              T6500300
              T6500310
              T6500320
              T6500330
              T6500340
              T6500350
              T6500360
              T6500370
              T6500380
              T6500390

SUBROUTINE EXPANC
  DIMENSION A( 7,7 ),AC(25),C(112,112),CCX(20,21),CCY(20,21),
  1FF(25),FFF(25),FL(25),FM(25),FR(3,50),FW(25),NP(112),NC(21),
  3P(112),PHI(25,20),PHIA(25,112),
  COMMON      A      ,AC      ,ACCEL      ,DT2      ,FFF      ,FREQ      ,BETA      ,
  1C      ,CCX      ,FL      ,FM      ,FR      ,
  2FF      ,
  3FW      ,
  7ICOD      ,IND      ,IRWS      ,ITAB      ,
  8JCCD      ,JCOD      ,JRWS      ,
  9KL,KK      ,KPRS      ,KRWS      ,LCOD      ,MCD      ,KC      ,MRWS      ,IMS      ,KLNS      ,
  COMMON      NC      ,NCLSL      ,NIVLS      ,NP      ,NPRQ      ,NPRS      ,
  INRWS      ,PHIA      ,PHI      ,PX      ,PY      ,
  2P      ,TM1      ,TM2

  COMMON EXTRA(10)
  DO 75 I=1,NCLS
  CALL TLU2(NP(I),PX(1,I),PY(1,I),FREQ,P(I),IND)
  P(I) = P(I)**.5
  IF(IND)70,75,70
  ITAB = ITAB+1
  CONTINUE
  I1 = 1
  I2 = KL
  KKK = NCLS/KL
  DO 200 N=1,KKK

```

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T6500400
 T6500410
 T6500420
 T6500430
 T6500440
 T6500450
 T6500460
 T6500470
 T6500480
 T6500490
 T6500500
 T6500510
 T6500520
 T6500530
 T6500540
 T6500550
 T6500560
 T6500570
 T6500580
 T6500590
 T6500600
 T6500610
 T6500620
 T6500630
 T6500640
 T6500650
 T6500660
 T6500670
 T6500680
 T6500690
 T6500700
 T6500710
 T6500720
 T6500730
 T6500740
 T6500750
 T6500760
 T6500770
 T6500780
 T6500790

```

DO 100 I=I1,I2
DO 100 J=I ,I2
L = I-(N-I)*KL
M = J-(N-I)*KL
100 C(I,J) = P(I)*A(L,M)*P(J)
I1 = I2+1
200 I2 = I2+KL
RETURN
END

$IBFTC DPRT      REF
CDPRT            PRINT DIAGONAL MATRIX (DPRT)

SUBROUTINE DPRT (N,A,NL)
DIMENSION A(25)
DIMENSION A(25)
NCOL=1
1   KCLS=N-NCOL+1
IF (KCLS-8) 100,101,101
100 KCOL=NCOL+KCLS-1
GO TO 102
101 KCOL=NCOL+7
102 IF (54-NL) 103,103,104
103 WRITE (6,200)
NL=0
104 NL=NL+2
WRITE (6,201)((K,K),K=NCOL,KCOL)
WRITE (6,202)(A(K),K=NCOL,KCOL)
NCOL=KCOL+1
IF (KCLS-8) 105,105,1
105 CONTINUE
RETURN
200 FORMAT (1H1)
201 FORMAT (6X,8(2X,I3,2X,I3,4X))
202 FORMAT (6X,1P8E14.7)
END

$IBFTC DTRM      REF
CDTRM            DIAGONAL MATRIX PRODUCT ROUTINE (DTRM)

SUBROUTINE DTRM (N,M,A,P,R)
DIMENSION A(25),P(25,20),R(20)
DIMENSION A(25),P(25,20),R(20)
DO 100 I=1,M
    
```

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```

R(I)=0.0
DO 100 J=1,N
  R(I)=R(I)+P(J,I)*A(J)
RETURN
END
$IBFTC TLU2      REF
CTLU2           LINEAR INTERPOLATION ROUTINE (TLU2)
SUBROUTINE TLU2 (N,XV,YV,X,Y,IND)
  DIMENSION XV(20),YV(20)
  DIMENSION XV(20),YV(20)
  IND = 0
  IF (N-1) 40,40,10
  10 IF (XV(1)-X) 20,60,40
  20 DO 30 I=2,N
    IF (XV(I)-X) 30,70,50
  30 CONTINUE
  40 IND = 1
  GO TO 80
  50 Y = (X-XV(I-1))*(YV(I)-YV(I-1))/(XV(I)-XV(I-1))+YV(I-1)
  GO TO 90
  60 I = 1
  70 Y = YV(I)
  GO TO 90
  80 Y = 0.0
  90 RETURN
END
$ORIGIN          ALPHA
$IBFTC PART1     FULIST,REF
SUBROUTINE PART1
CPART1
  DIMENSION A( 7,7 ),AC(25),C(112,112),CCX(20,21),CCY(20,21),
  1FF(25),FFF(25),FL(25),FM(25),FMD(3,25),
  2HD(10),NP(112),NC(21),
  3P(112),PHI(25,20),PHIA(25,112),
  4SIT(112),S2T(112),SVAL(20),TINT(112)
  DIMENSION      XA(25,31),  XB(25,30),
  1XD(25,30),     XKM(25),    XKN(112),
  2XLN(112),      XM(25,30),  XN(25,30),
  3XSUM2(112)
  DIMENSION NLN(25),NCN(25)
  FR(3,50),FW(25),
  PX(20,112),PY(20,112),
  XC(25,31),
  XLM(25),
  XSUM1(25),

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COMMON	ACCEL	BETA
1C	DT1	
2FF	DT2	
3FW	FR	
7ICOD	IRWS	IMS
8JCCD	JRWS	KLNS
9KLKK	KRWS	KC
COMMON	LCOD	MRWS
1NRWS	NCLVLS	NPRQ
2P	PHI	PY
	TM1	
	TM2	
COMMON EXTRA(10)		
EQUIVALENCE (C(1),XA(1)),(C(776),XB(1)),(C(1526),XC(1)),		
1(C(2301),XD(1)),(C(3051),XKM(1)),(C(3076),XKN(1)),		
2(C(3188),XLM(1)),(C(3213),XLN(1)),(C(3325),XM(1)),		
3(C(4075),XN(1)),(C(4825),XSUM1(1)),(C(4850),XSUM2(1))		
EQUIVALENCE (C(5962),NLN(1)),(C(5990),NCN(1))		
NPRQ = 1		

COMMON	ACCEL	BETA
1C	DT1	
2FF	DT2	
3FW	FR	
7ICOD	IRWS	IMS
8JCCD	JRWS	KLNS
9KLKK	KRWS	KC
COMMON	LCOD	MRWS
1NRWS	NCLVLS	NPRQ
2P	PHI	PY
	TM1	
	TM2	
COMMON EXTRA(10)		
EQUIVALENCE (C(1),XA(1)),(C(776),XB(1)),(C(1526),XC(1)),		
1(C(2301),XD(1)),(C(3051),XKM(1)),(C(3076),XKN(1)),		
2(C(3188),XLM(1)),(C(3213),XLN(1)),(C(3325),XM(1)),		
3(C(4075),XN(1)),(C(4825),XSUM1(1)),(C(4850),XSUM2(1))		
EQUIVALENCE (C(5962),NLN(1)),(C(5990),NCN(1))		
NPRQ = 1		

FREQUENCY DATA INPUT

CALL CLOCK (TM1, TM2, DT1, DT2)

READ (5, 2012) NCASE, HD

WRITE (6, 1000) DT1, DT2

WRITE (6, 2013) NCASE, HD

READ (5, 2006) NIVLS

WRITE (6, 1001) NIVLS

DO 1 J=1, NIVLS

READ (5, 2001) (FR(I, J), I=1, 3)

WRITE (6, 1002) (FR(I, J), I=1, 3)

PRGM C PRIME

READ(5, 4000) ICOD, JCOD, ACCEL, HD

WRITE(6, 4001) ICOD, JCOD, ACCEL, HD

JCCD = JCOD+1

K=ICOD/2+JCOD/2

IF(ICOD-3) 9, 2015, 2015

IF(K) 48, 48, 10

ALL FLOATING POINT VARIABLES START WITH X
REVISION IS AN OPTION TO USE A MORE COMPLICATED
FUNCTION TO CALCULATE THE PHIA MATRIX

2015 READ (5, 2085) NT, NS, XXR, XL1, XDELTA, XL

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T6501600
T6501610
T6501620
T6501630
T6501640
T6501650
T6501660
T6501670
T6501680
T6501690
T6501700
T6501710
T6501720
T6501730
T6501740
T6501750
T6501760
T6501770
T6501780
T6501790
T6501800
T6501810
T6501820
T6501830
T6501840
T6501850
T6501860
T6501870
T6501880
T6501890
T6501900
T6501910
T6501920
T6501930
T6501940
T6501950
T6501960
T6501970
T6501980
T6501990

```

2085 FORMAT(6X,2I3,3X,4E15.6)
2086 FORMAT(4E15.6)
10001 FORMAT(1H0,10X,27HINPUT DATA FOR FOURIER CALC///
111X, 5HNT = ,I3/
211X, 5HNX = ,I3/
311X, 4HR = ,1PE15.6/
411X, 4HL = ,1PE15.6/
511X, 7HLBAR = ,1PE15.6/
611X, 7HGAMMA = ,1PE15.6/
711X, 6HNRWS = ,I3)

10002 FORMAT(1H1,10X, 5HNL = I3,10X, 7HMODE = I3/
119X,1H1,16X,1HA,24X,1HB,24X,1HM)
10003 FORMAT(18X,I2,9X,3(1PE15.6,10X))
10004 FORMAT(1H1,10X, 5HNC = I3,10X,7HMODE = I3/
119X,1H1,16X,1HC,24X,1HD,24X,1HN)
READ (5,2085)NRWS
WRITE (6,10001)NT,NS,XXR,XL1,XL,XDELTA,NRWS
NCLS = NT*NS
DO 12015 N=1,NRWS
READ (5,2085)NNS
WRITE (6,10002)NNS,N
READ (5,2086)XA(N,1)
N1 = 0
WRITE (6,10003)N1,XA(N,1)
IF(NNS-1)12012,12012,12010
12010 DO 12011 I=2,NNS
READ (5,2086)XA(N,I),XB(N,I-1),XM(N,I-1)
N1 = N1+1
12011 WRITE (6,10003)N1,XA(N,I),XB(N,I-1),XM(N,I-1)
12012 NLN(N)=NNS
READ (5,2085)NNTH
WRITE (6,10004)NNTH,N
READ (5,2086)XC(N,1)
N1 = 0
WRITE (6,10003)N1,XC(N,1)
IF(NNTH-1)12015,12015,12013
12013 DO 12014 I=2,NNTH
READ (5,2086)XC(N,I),XD(N,I-1),XN(N,I-1)
N1 = N1+1
12014 WRITE (6,10003)N1,XC(N,I),XD(N,I-1),XN(N,I-1)

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12015 NCN(N) = NNTH
      XNT = NT
      XNS = NS
      XDTH = 360.0/(XNT*57.2957795)
      XDELS = XL1/XNS
      DO12090 IR=1,NRWS
      NNS = NLN(IR)
      NNTH = NCN(IR)
      NNS1 = NNS-1
      NNTH1 = NNTH-1
      DO 2016 I=1,NNS1
      XKM(I) = COS(XM(IR,I)/XL*XDELS*3.1415926)-1.0
      XLM(I) = SIN(XM(IR,I)/XL*XDELS*3.1415926)
2016 DO 2017 I=1,NNTH1
      XKN(I) = COS(XN(IR,I)*XDTH)-1.0
      XLN(I) = SIN(XN(IR,I)*XDTH)
2017 DO 2040 I=1,NS
      XSUM = 0.0
      XTIME = I-1
      XSU = XTIME*XDELS
      IF(XDELTA)12017,12019,12019
12017 WRITE (6,12018)
12018 FORMAT(33H1PROGRAM CANNOT CONTINUE GAMMA = F12.4)
      CALL EXIT
12019 DO 2020 J=1,NNS1
      XANGLE = XM(IR,J)*3.1415926*XDELTA/XL
      XAM = XA(IR,J+1)*COS(XANGLE)+XB(IR,J)*SIN(XANGLE)
      XBM = -XA(IR,J+1)*SIN(XANGLE)+XB(IR,J)*COS(XANGLE)
2020 XSUM = XSUM+1.0/XM(IR,J)*(XAM*(XKM(J)*SIN(XM(IR,J)/XL*XSU*3.1415926)
      126)+XLM(J)*COS(XM(IR,J)/XL*XSU*3.1415926) ) -XBM*(XKM(J)*COS(XM(IR)
      2,J)/XL*XSU*3.1415926) -XLM(J)*SIN(XM(IR,J)/XL*XSU*3.1415926))
2040 XSUM1(I) = (XSUM+XA(IR,1)*XDELS*3.1415926*.5/XL)*XL/3.1415926*XXR
      DO 2060 I=1,NT
      XSUM = 0.0
      XTIME = I-1
      XTETHA = XTIME*XDTH
      DO 2050 J=1,NNTH1
2050 XSUM = XSUM+1.0/XN(IR,J)*(XC(IR,J+1)*(XKN(J)*SIN(XN(IR,J)*XTHETA)
      1+XLN(J)*COS(XN(IR,J)*XTHETA)) -XD(IR,J)*(XKN(J)*COS(XN(IR,J)*XTHET
      2A)- XLN(J)*SIN(XN(IR,J)*XTHETA))

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T6502400
T6502410
T6502420
T6502430
T6502440
T6502450
T6502460
T6502470
T6502480
T6502490
T6502500
T6502510
T6502520
T6502530
T6502540
T6502550
T6502560
T6502570
T6502580
T6502590
T6502600
T6502610
T6502620
T6502630
T6502640
T6502650
T6502660
T6502670
T6502680
T6502690
T6502700
T6502710
T6502720
T6502730
T6502740
T6502750
T6502760
T6502770
T6502780
T6502790

```

2060      XSUM2(I) = XSUM+XC(IR,1)*XDTH*.5
        IK = 0
        DO 2080 J=1,NT
        DO 2080 I=1,NS
        IK = IK+1
2080      PHIA(IR,IK) = XSUM1(I)*XSUM2(J)
12090     CONTINUE
        GO TO 51

C          READ BASIC INPUT DATA
10      READ (5,3001)FALP
        READ (5,3000)NRWS
        READ(5,3003)((FMD(I,J),I=1,3),J=1,NRWS)
        WRITE (6,4002)FALP
        WRITE (6,4003)NRWS
        WRITE(6,4004)((FMD(I,J),I=1,3),J=1,NRWS)
        X=FALP/57.2957795
        CALP=COS(X)
        SALP=SIN(X)
        CSQ=CALP**2
        READ (5,3001)S1,S2
        WRITE (6,4005)S1,S2
        FLNG=S2-S1
        CFLMT=FLNG/3.14159265
        TH1=0.0
        TH2=360.0

C          CHECK CALC OF PHIAT
11      GO TO(12,11,12),ICOD
        READ (5,3000)NDUM,NCLS
        READ (5,3000)NT,NS
        WRITE(6,4006)ICOD,NCLS,NT,NS

C          CALC BASIC PARAMETERS
        FLTT=NT
        FLTS=NS
        DTH=(TH2-TH1)/FLTT
        DS=FLNG/FLTS

C          CHECK CALC OF PHI2T
12      GO TO (14,14,13), JCCD
13      READ (5,3000)NDUM,NPRS
        WRITE(6,4007)JCOD,NPRS
        READ (5,3001)(SVAL(I),I=1,NPRS)

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C      WRITE (6,3002)(SVAL(I),I=1,NPRS)
      CYCLE ON MODE COUNTER
14     DO 42 II=1,NRWS
        FLIN=FMD(1,II)
        FLIM=FMD(2,II)
        FLTA=FMD(3,II)
        FLMT=CFLMT/FLTM
      C      GO TO(39,17,39),ICOD
      C      CHECK CALC OF PHIAT
      C      CALC PHIAT
      C      CALC INTEGRALS OF THETA FUNCTION
17     I=1
        THTT=THT1
        IF (FLIN) 18,18,19
        FX2=THTT/57.2957795
        GO TO 20
18     X=FLIN*THTT/57.2957795
        FX2=SIN(X)
        THTT=THTT+DTHT
20     IF (ABS(THTT-THT2)-1.0) 22,22,21
        IF (THTT-THT2) 23,23,22
        THTT=THT2
        FX1=FX2
        IF (FLIN) 24,24,25
        FX2=THTT/57.2957795
        GO TO 26
24     X=FLIN*THTT/57.2957795
        FX2=SIN(X)
        TINT(I)=FX2-FX1
26     IF (THTT-THT2) 27,28,28
        I=I+1
        GO TO 20
      C      CALC INTEGRALS OF X FUNCTION
28     J=1
        STOT=S1
        X=STOT/FLMT
        XQ=X**2
        CA=SIN(X)
        CB=COS(X)
        F12=-XQ*CB+2.0*X*CA+2.0*CB

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T6503200
T6503210
T6503220
T6503230
T6503240
T6503250
T6503260
T6503270
T6503280
T6503290
T6503300
T6503310
T6503320
T6503330
T6503340
T6503350
T6503360
T6503370
T6503380
T6503390
T6503400
T6503410
T6503420
T6503430
T6503440
T6503450
T6503460
T6503470
T6503480
T6503490
T6503500
T6503510
T6503520
T6503530
T6503540
T6503550
T6503560
T6503570
T6503580
T6503590

```

29      F22= XQ*CA+2.0*X*CB-2.0*CA
      STOT=STOT+DS
30      IF (ABS(STOT-S2)-1.0) 31,31,30
31      IF (STOT-S2) 32,32,31
32      STOT=S2
      F11=F12
      F21=F22
      X=STOT/FLMT
      XQ=X**2
      CA=SIN(X)
      CB=COS(X)
      F12=-XQ*CB+2.0*X*CA+2.0*CB
      F22= XQ*CA+2.0*X*CB-2.0*CA
      S1T(J)=F12-F11
      S2T(J)=F22-F21
      IF (STOT-S2) 33,34,34
33      J=J+1
      GO TO 29
34      FLMT=FLMT**3
      S1C=FLMC*COS(S1/FLMT)
      S2C=FLMC*SIN(S1/FLMT)
      IF (FLTN) 35,35,36
35      CBTA=CSQ*FLTA
      GO TO 37
36      CBTA=(CSQ*FLTA)/FLTN
37      JJ=1
      DO 38 KK=1,NT
      BETA=CBTA*TINT(KK)
      DO 38 LL=1,NS
      PHIA(II,JJ)=BETA*(S1C*S1T(LL)-S2C*S2T(LL))
      JJ=JJ+1
38      C
      CHECK CALC OF PHI2T
39      GO TO (42,42,40),JCCD
      C
      CALC PHI2T
40      DO 41 JJ=1,NPRS
      X=(SVAL(JJ)-S1)/FLMT
      PHIA(II,JJ)=(FLTA*SVAL(JJ)*CALP)**2*(SIN(X))**2
41      CONTINUE
42      C
      C
      C
      MATRIX COMPUTATION IS DONE

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```

IRWS = NRWS
JRWS=NRWS
KRWS=NRWS
KPRS=NSTS
MRWS=NRWS
RETURN
1000 FORMAT(1H110X44HSHELL RESPONSE TO RANDOM PRESSURE EXCITATION
      111X,10HNAS 1-3179,3X,A6,A2/)
1001 FORMAT (1H0,10X,20HNO. OF CALC. FREQ. =,I4/1H0,19X,
      19HCALC FREQ,12X,13HCALC INTERVAL,8X,23HNO. CALC / NO. PRINTOUT)
1002 FORMAT(11X3(1PE21.6))
2001 FORMAT (1P8E15.7)
2006 FORMAT (6X,3I3)
2012 FORMAT (6X,I3,3X,10A6)
2013 FORMAT (9X,I3,3X,10A6/)
3000 FORMAT (6X,4I3)
3001 FORMAT (1P4E15.7)
3002 FORMAT (1H0,1P8E15.5)
3003 FORMAT(2E10.4,E15.7)
4000 FORMAT(3X,2I3,3X,1PE10.4,10A5)
4001 FORMAT(1H110X,
      118HPHIAT CALC CODE = ,I2,6X,18HPHI2T CALC CODE = ,I2/
      11X,16HACCEL. SCALAR = ,
      31PE10.4/11X,10A5/)
4002 FORMAT (1H0,10X,8HALPHA = ,1PE15.6)
4003 FORMAT (1H0,10X,8HMODES = ,I3//1H0,17X,
      11HN,13X,1HM,11X,3HAMN)
4004 FORMAT(9X,1P2E14.4,1PE14.6)
4005 FORMAT (1H0,10X,5HX1 = ,1PE14.6,3X,5HXE = ,1PE14.6)
4006 FORMAT(1H010X12HPHIAT CODE =I2,12H N COLS =I3,
      18X,4HNT =,I3,8X,4HNX =,I3)
4007 FORMAT(1H010X12HPHI2T CODE =I2,12H N COLS =I3//
      *10X21HPHI2T RESPONSE POINTS)
END
$ORIGIN ALPHA
$IBFTC FHALF FULIST,REF
SUBROUTINE FHALF
CFHALF
      DIMENSION A( 7,7 ),AC(25),C(112,112),CCX(20,21),CCY(20,21),
      1FF(25),FFF(25),FL(25),FM(25),FR(3,50),FW(25),NP(112),NC(21),

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3P(112),PHI(25,20),PHIA(25,112),
COMMON A ,AC ,ACCEL ,DT2 ,BETA ,
1C ,CCX ,CCY ,DT1 ,FFF ,
2FF ,FL ,FLNG ,FM ,FR ,FREQ ,
3FW , , , ,IMS ,
7ICOD ,IND ,IRWS ,ITAB , ,KLNS ,
8JCCD ,JCDD ,JRWS , ,MRWS ,
9KL,KK ,KPRS ,KRWS ,LCOD ,MCOD ,
COMMON NC ,NCLS ,NIVLS ,NP ,NPRS ,
1NRWS ,PHIA ,PHI ,PX ,PY ,
2P ,TM1 ,TM2 ,
COMMON EXTRA(10)
DATA Q000CT/O473031216360/
PHA =Q000CT
DATA Q001CT/O473031026360/
PHI2=Q001CT
DATA Q002CT/O473062026360/
PHS2=Q002CT
GO TO(101,125,125),ICOD
48 C PHIA1 INPUT DIRECTLY
101 READ(5,2006)IRWS,NCLS
NRWS=IRWS
MRWS = IRWS
DO 102 I=1,IRWS
102 READ (5,2005)(PHIA(I,J),J=1,NCLS)
C PHIA1 PRINTOUT
125 KLNS=NCLS/8+1
NLNS=54
DO 128 I=1,IRWS
IF (54-NLNS-KLNS) 126,126,127
126 WRITE (6,1000)DT1,DT2
WRITE(6,1013)ICOD,IRWS,NCLS,PHA
NLNS=5
NLNS=NLNS+KLNS
127 WRITE (6,2006)I
WRITE (6,2001)(PHIA(I,J),J=1,NCLS)
128 GO TO ( , 129,150),JCUD
C PHIA2 INPUT DIRECTLY
129 READ(5,2006)JRWS,NPRS
DO 132 I=1,JRWS

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132 READ (5,2005)(PHI(I,J),J=1,NPRS)
C      PHI2T PRINTOUT
150   KLNS=NPRS/8+1
      NLNS=54
      DO 153 I=1,JRWS
      IF (54-NLNS-KLNS) 151,151,152
151   WRITE (6,1000)DT1,DT2
      WRITE(6,1013)JCOD,JRWS,NPRS,PHI2
      NLNS=5
152   NLNS=NLNS+KLNS
      WRITE (6,2006)I
153   WRITE (6,2001)(PHI(I,J),J=1,NPRS)
      IF (IRWS-JRWS) 1530,1786,1530
1530  WRITE (6,1042)IRWS,JRWS
      GO TO 999

C      INPUT SPECTRUM DATA
1786  WRITE (6,1000)DT1,DT2
2010  READ(5,2010)LCOD,MCOD,EXTRA(6),FAC1,FAC2
      FORMAT(2I1,F1.0,2F15.0)
      WRITE (6,1020)LCOD,MCOD,FAC1,FAC2
C      INPUT PSD TABLES
17861 WRITE (6,10211)KL
      GO TO 1789
1788  WRITE (6,1021)KL,KC,KK
1789  IF (EXTRA(6))30000,30000,30005
30000 WRITE(6,30001)
30001 FORMAT (1H0,10X,27HCONSTANT CORRELATION METHOD)
      GO TO 31789
30005 READ(5,30006)(EXTRA(I),I=1,5)
30006 FORMAT(5F15.0)
      WRITE(6,30002)(EXTRA(I),I=1,5)
30002 FORMAT (1H0,10X,27HEMPIRICAL CO-SPECTRA METHOD)11X,
127HLONGITUDINAL DECAY RATE E15.8/11X
127HCIRCUMFERENTIAL DECAY RATE E15.8/11X
127HAVERAGE SIZE OF AREA(LONG.)E15.8/11X
127HAVERAGE SIZE OF AREA(CIR.) E15.8/11X
119HCONVECTION VELOCITY8XE15.8)
31789 GO TO(179,1790,3789),LCOD

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C 179 N=KL
    GO TO 1791

C 1790 N=1
    WRITE (6,1045)
    GO TO 1791

C 3789 N = NCLS
    1791 DO 181 I=1,N
        READ (5,2006) NP(I)
        WRITE (6,1022) I, I, NP(I)
        J=NP(I)
        READ (5,2004) (PX(K,I),K=1,J)
        READ (5,2004) (PY(K,I),K=1,J)
        DO 180 K=1,J
            PX(K,I)=FAC1*PX(K,I)
            PY(K,I)=FAC2*PY(K,I)
            WRITE (6,1025) (PX(K,I),K=1,J)
            WRITE (6,1027) (PY(K,I),K=1,J)
            IF (LCOD-3) 182,3793,182
            IF (ICOD-2) 3794,3795,3794
            IF (JCOD-2) 182,3795,182
            WRITE (6,3796)
        FORMAT(55H1 RESPONSE CALCULATION INVOLVING PHI2T AND PHIAT MATRIX/
        50H DEPENDING UPON ASSUMPTION OF SYMMETRIC PRESSURE
        30H POWER SPECTRA DISTRIBUTION.
        47H LCOD = 3 INDICATES UNSYMMETRIC DISTRIBUTION
        INPUT CO-SPECTRUM TABLES

C 182 WRITE (6,1000) DT1,DT2
    NXCO = 0
    IF (MCO) 1850,1830,1850

    1830 DO 185 J=1,KK
        DO 185 I=1,KC
            NXCO = NXCO+1
            READ (5,2006) NC(NXCO)
            IF (NC(NXCO)-20) 1831,1831,1832
            1832 WRITE (6,1024) I,J
            CCX(1,NXCO) = 0.
            CCX(2,NXCO) = 9999.

T6504800
T6504810
T6504820
T6504830
T6504840
T6504850
T6504860
T6504870
T6504880
T6504890
T6504900
T6504910
T6504920
T6504930
T6504940
T6504950
T6504960
T6504970
T6504980
T6504990
T6505000
T6505010
T6505020
T6505030
T6505040
T6505050
T6505060
T6505070
T6505080
T6505090
T6505100
T6505110
T6505120
T6505130
T6505140
T6505150
T6505160
T6505170
T6505180
T6505190
    
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T6505200
T6505210
T6505220
T6505230
T6505240
T6505250
T6505260
T6505270
T6505280
T6505290
T6505300
T6505310
T6505320
T6505330
T6505340
T6505350
T6505360
T6505370
T6505380
T6505390
T6505400
T6505410
T6505420
T6505430
T6505440
T6505450
T6505460
T6505470
T6505480
T6505490
T6505500
T6505510
T6505520
T6505530
T6505540
T6505550
T6505560
T6505570
T6505580
T6505590

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CCY(1,NXCO) = 9999.
CCY(2,NXCO) = 9999.
GO TO 185
1831 WRITE(6,1023)I,J,NC(NXCO)
      K = NC(NXCO)
      READ (5,2004)(CCX(L,NXCO),L=1,K)
      READ (5,2004)(CCY(L,NXCO),L=1,K)
      DO 184 L=1,K
        CCX(L,NXCO) = CCX(L,NXCO)*FAC1
        WRITE (6,1025)(CCX(L,NXCO),L=1,K)
        WRITE(6,1026)(CCY(L,NXCO),L=1,K)
185  CONTINUE
      GO TO 1854
1850 DO 1853 I=1,KL
      DO 1853 J=1,KL
        IF (I-J) 1851,1853,1851
        NXCO = NXCO+1
        READ (5,2006)NC(NXCO)
        IF(NC(NXCO)-20)2851,2851,2852
2852  WRITE(6,1024)I,J
        CCX(1,NXCO) = 0.
        CCX(2,NXCO) = 9999.
        CCY(1,NXCO) = 9999.
        CCY(2,NXCO) = 9999.
      GO TO 1853
2851  WRITE(6,1023)I,J,NC(NXCO)
      K = NC(NXCO)
      READ (5,2004)(CCX(L,NXCO),L=1,K)
      READ (5,2004)(CCY(L,NXCO),L=1,K)
      DO 1852 L=1,K
        CCX(L,NXCO) = CCX(L,NXCO)*FAC1
        WRITE (6,1025)(CCX(L,NXCO),L=1,K)
        WRITE (6,1026)(CCY(L,NXCO),L=1,K)
1853  CONTINUE
      C
1854  WRITE (6,1000)DT1,DT2
      READ (5,2006)N
      READ (5,2002)(FM(I),FL(I),FF(I),I=1,N)
      WRITE (6,1031)N
      WRITE (6,2009)(FM(I),FL(I),FF(I),I=1,N)

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INPUT NORMAL MODE PARAMETERS

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186 IF (NRWS-N) 186,187,186
    WRITE (6,1046)NRWS,N
    GO TO 999
187 DO 188 I=1,N
188 FW(I)=6.28318531*FF(I)
    J=0
    DO 191 I=1,N
    IF (FR(1,I)-FF(I)) 189,191,191
189 IF (FR(1,NIVLS)-FF(I)) 191,191,190
190 FFF(J+1)=FF(I)
    J=J+1
191 CONTINUE
    IMS=J
    DO 193 I=1,IMS
    DO 193 J=1,IMS
    IF (FFF(I)-FFF(J)) 193,194,192
192 FFFI=FFF(I)
    FFF(I)=FFF(J)
    FFF(J)=FFFI
    GO TO 193
194 IF (I-J) 195,193,195
195 FFF(J)=1.E8
193 CONTINUE
    RETURN
999 CONTINUE
9990 CALL EXIT
    RETURN
1000 FORMAT(1H110X44HSHELL RESPONSE TO RANDOM PRESSURE EXCITATION
    111X,10HNAS 1-3179,3X,A6,A2/)
1013 FORMAT(1H010X6HCODE =I4,3X3HN =I4,3X3HM =I4,3XA6)
1020 FORMAT(1H010X10HPSD CODE =I4,3X15HCO-SPEC. CODE =I4,3X
    113HFREQ. MULT. =1PE15.6,3X11HPSD MULT. =1PE15.6)
1021 FORMAT (1H0,10X,5HKL = ,I3,3X,5HKL = ,I3/)
10211 FORMAT (1H0,10X,5HKL = ,I3)
1022 FORMAT(1H0,10X15HPSD TBLF, ELF (I3,1H,I3,6H) WITHI3,5H PTS.)
1023 FORMAT (1H0,10X,15HCO- TBLF, ELF (,I1,1H,,I1,6H) WITH,I3,5H PTS.)
1024 FORMAT(1H010X15HCO- TBLF, ELF (I1,1H,I1,18H) TO BE CALCULATED)
1025 FORMAT(4X,5HFREQ./10(2X1PE10.3))
1026 FORMAT(4X,8HCO-SPEC./10(2X1PE10.3))
1027 FORMAT(4X,3HPSD/10(2X1PE10.3))
  
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T6505600
 T6505610
 T6505620
 T6505630
 T6505640
 T6505650
 T6505660
 T6505670
 T6505680
 T6505690
 T6505700
 T6505710
 T6505720
 T6505730
 T6505740
 T6505750
 T6505760
 T6505770
 T6505780
 T6505790
 T6505800
 T6505810
 T6505820
 T6505830
 T6505840
 T6505850
 T6505860
 /T6505870
 T6505880
 T6505890
 T6505900
 T6505910
 T6505920
 T6505930
 T6505940
 T6505950
 T6505960
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1031 FORMAT (1H0,10X,11HNO MODES = ,I3/
      112X,12H GEN. MASSES,3X,12H DMP. COEFFS,3X,12H MODAL FREQS/)
1042 FORMAT (1H0,10X,34HERROR, PHIT AND PHIAT,INCOMPATIBLE,2I4)
1045 FORMAT (1H0,10X,32HPSD IS SCALAR FROM 1 TABLE INPUT)
1046 FORMAT (1H0,10X,36HERROR,PHIAT AND NORMAL MODES UNEQUAL,2I3)
2001 FORMAT (1P8E15.7)
2002 FORMAT (1P3E15.7)
2004 FORMAT (1P4E15.7)
2005 FORMAT (12X,1P4E15.7)
2006 FORMAT (6X,3I3)
2008 FORMAT (6X,2I3,1P2E15.7)
2009 FORMAT (10X,1P3E15.6)
      END
$ORIGIN          ALPHA
$IBFTC SHALF    FULIST,REF
      SUBROUTINE SHALF
C$HALF
      DIMENSION A( 7,7 ),AC(25),C(112,112),CCX(20,21),CCY(20,21),
      1FF(25),FFF(25),FL(25),FM(25),FR(3,50),FW(25),NP(112),NC(21),
      3P(112),PHI(25,20),PHIA(25,112),
      4R(25),R1(25),T(25),TT(25),SC(25)
      COMMON      A      ,AC      ,ACCEL      ,DT2      ,FFF      ,
      1C      ,CCX      ,CCY      ,FM      ,FR      ,FREQ      ,
      2FF      ,FL      ,FLNG      ,IRWS      ,ITAB      ,IMS      ,
      3FW      ,IND      ,JCOD      ,JRWS      ,KCRS      ,KLNS      ,
      7ICOD      ,JCOD      ,KPRS      ,LCOD      ,MCD      ,MRWS      ,
      8JCCD      ,KK      ,KPRS      ,NCLS      ,NIVLS      ,NPRQ      ,
      9KL,KK      ,KPRS      ,NC      ,PHIA      ,PHI      ,PY      ,
      COMMON      NC      ,PHIA      ,PHI      ,TM1      ,TM2      ,
      1NRWS      ,PHIA      ,PHI      ,TM1      ,TM2      ,
      2P      ,TM1      ,TM2      ,
      COMMON EXTRA(10)
      INITIALIZE FREQUENCY CYCLE
      IPS=1
      INS=1
      DO 196 I=1,NRWS
      R(I) = 0.0
      R1(I) = 0.0
      T(I) = 0.0
      TT(I) = 0.0
196      TT(I) = 0.0
      T6506000
      T6506010
      T6506020
      T6506030
      T6506040
      T6506050
      T6506060
      T6506070
      T6506080
      T6506090
      T6506100
      T6506110
      T6506120
      T6506130
      T6506140
      T6506150
      T6506160
      T6506170
      T6506180
      T6506190
      T6506200
      T6506210
      T6506220
      T6506230
      T6506240
      T6506250
      T6506260
      T6506270
      T6506280
      T6506290
      T6506300
      T6506310
      T6506320
      T6506330
      T6506340
      T6506350
      T6506360
      T6506370
      T6506380
      T6506390

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T6506400
T6506410
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T6506430
T6506440
T6506450
T6506460
T6506470
T6506480
T6506490
T6506500
T6506510
T6506520
T6506530
T6506540
T6506550
T6506560
T6506570
T6506580
T6506590
T6506600
T6506610
T6506620
T6506630
T6506640
T6506650
T6506660
T6506670
T6506680
T6506690
T6506700
T6506710
T6506720
T6506730
T6506740
T6506750
T6506760
T6506770
T6506780
T6506790

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NVAL=0
NIVL=1
FREQ=FR(1,NIVL)
DFRQ=FR(2,NIVL)
KCNT=FR(3,NIVL)+0.0000001
KPRC=KCNT
X1 = EXTRA(1)*EXTRA(3)
X2 = EXTRA(2)*EXTRA(4)

      BEGINNING OF CYCLE

C 200 ITAB=0
      GO TO(20151,205,20101),LCOD
20101 IF(MCOD)20121,20121,207
20121 LM1 = NCLS
      GO TO 201
20151 LM1 = KL
C
201 DO 203 I=1,LM1
      CALL TLU2 (NP(I),PX(1,I),PY(1,I),FREQ,P(I),IND)
      P(I) = P(I)**.5
      IF (IND) 202,203,202
202 ITAB=ITAB+1
203 CONTINUE
      IF(LCOD-2)2019,2019,207
2019 DO 204 J=1,NCLS,KL
      DO 204 I=1,KL
      II=J+I-1
      P(II)=P(I)
204 GO TO 207
C
205 CALL TLU2 (NP,PX,PY,FREQ,SCAL,IND)
      IF (IND) 206,207,206
206 ITAB=ITAB+1
C
207 OMEGA = 3.1416*EXTRA(3)*FREQ/(6.*EXTRA(5))
      IF(MCOD)2090,2070,2090
2070 ICOUN = 0
      DO 209 J=1,KK
      DO 209 I=1,KC
      ICOUN = ICOUN+1
      CALL TLU2(NC(ICOUN),CCX(1,ICOUN),CCY(1,ICOUN),FREQ,A(I,J),IND)

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208 IF (IND) 208,209,208
209 ITAB=ITAB+1
209 CONTINUE
30025 IF(EXTRA(6))2095,2095,30025
30025 A(1,1) = 9999.
DO 30075 J=1,KK
XX3 = J
DO 30075 I=1,KC
XX2 = I
IF (I-1) 30030,30030,30035
30030 XX1= 1.
GO TO 30040
30035 XX1= 0.
30040 IF (A(I,J).NE.9999.) GO TO 30044
XCC1 = (X1**2+OMEGA**2)**2
XCC2 = (X1**2-OMEGA**2)
XCC3 = EXP(-X1*XX2)*COS(OMEGA*XX2)+EXP(-X1*(XX2-2.))*
1 COS(OMEGA*(XX2-2.))-2.*EXP(-X1*(XX2-1.))*COS(OMEGA*(XX2-1.))
XCC4 = 0.
IF (XX1.EQ.0.) GO TO 30042
XCC4 = X1*OMEGA*(EXP(X1)-EXP(-X1))*SIN(OMEGA)
30042 XCC5 = 1.
IF (X1.EQ.0.) GO TO 30043
XCC5=(EXP(-X2*XX3)+EXP(-X2*(XX3-2.))-2.*EXP(-X2*(XX3-1.)))/(X2**2)
30043 XXC = (1./XCC1)*(XCC2*XCC3+XCC4)*XCC5
A(I,J) = XXC
GO TO 30075
30044 A(I,J) = A(I,J)*ABS(A(1,1))
30075 CONTINUE
A(1,1) = ABS(A(1,1))
GO TO 2095
2090 ICOUN = 0
DO 2094 I=1,KL
DO 2094 J=1,KL
IF(I-J)2088,2091,2094
2091 A(I,J)=1.0
GO TO 2094
2088 ICOUN = ICOUN+1
2092 CALL TLU2(NC(ICOUN),CCX(1,ICOUN),CCY(1,ICOUN),FREQ,A(I,J),IND)
IF (IND) 2093,2094,2093

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2093 ITAB=ITAB+1
2094 CONTINUE
      IF(EXTRA(6))2095,2095,30080
30080 A(1,1) = 9999.
      DO 30150 I=1,KL
      DO 30150 J=1,KL
      XX2 = J-I+1
      XX1 = 0.
      IF (XX2.GT.1.) GO TO 30081
      XX1 = 1.
30081 IF (I.NE.J.OR.I.EQ.1) GO TO 30100
30090 A(I,J) = ABS(A(1,1))
      GO TO 30150
30100 IF (A(I,J).NE.9999.) GO TO 30120
      XCC1 = (X1**2+OMEGA**2)**2
      XCC2 = (X1**2-OMEGA**2)
      XCC3 = EXP(-X1*XX2)*COS(OMEGA*XX2)+EXP(-X1*(XX2-2.))*
1 COS(OMEGA*(XX2-2.))-2.*EXP(-X1*(XX2-1.))*COS(OMEGA*(XX2-1.))
      XCC4 = 0.
      IF (XX1.EQ.0.) GO TO 30101
      XCC4 = X1*OMEGA*(EXP(X1)-EXP(-X1))*SIN(OMEGA)
30101 XCC5 = 1.
      IF (X1.EQ.0.) GO TO 30102
      XCC5 = (EXP(-X2)+EXP(X2)-2.)/(X2**2)
30102 XXC = (1./XCC1)*(XCC2*XCC3+XCC4)*XCC5
      A(I,J) = XXC
      GO TO 30150
30120 A(I,J) = ABS(A(1,1))*A(I,J)
30150 CONTINUE
      A(1,1) = ABS(A(1,1))
2095 IF (ITAB) 210,211,210
210 WRITE (6,1032)ITAB
      GO TO 999
211 GO TO(214,212,3221),LCOD
3221 IF(MCOD)2140,2140,3222
3222 CALL EXPANC
      GO TO 3272
212 IF(MCOD)5212,5212,5300
5300 DO 5305 I=1,KL
      DO 5305 J=1,KL

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T6507200
T6507210
T6507220
T6507230
T6507240
T6507250
T6507260
T6507270
T6507280
T6507290
T6507300
T6507310
T6507320
T6507330
T6507340
T6507350
T6507360
T6507370
T6507380
T6507390
T6507400
T6507410
T6507420
T6507430
T6507440
T6507450
T6507460
T6507470
T6507480
T6507490
T6507500
T6507510
T6507520
T6507530
T6507540
T6507550
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T6507570
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T6507620
T6507630
T6507640
T6507650
T6507660
T6507670
T6507680
T6507690
T6507700
T6507710
T6507720
T6507730
T6507740
T6507750
T6507760
T6507770
T6507780
T6507790
T6507800
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T6507830
T6507840
T6507850
T6507860
T6507870
T6507880
T6507890
T6507900
T6507910
T6507920
T6507930
T6507940
T6507950
T6507960
T6507970
T6507980
T6507990

5305 A(I,J) = A(I,J)*SCAL
GO TO 12270
5212 DO 213 I=1,KL
DO 213 J=1,KK
213 A(I,J)=SCAL*A(I,J)
214 IF (MCOD) 2270,2140,2270
2140 KKK=0
IVK=NCLS
DO 217 JJJ=1,KK
DO 216 II=1,KL
IJ=0
K11=KKK+II
K12=IVK+II
DO 216 JJ=II,KL
IJ=IJ+1
K21=KKK+JJ
K22=IVK+JJ
215 IF (KKK) 215,216,215
C(IJ,K12)=A(IJ,JJJ)
C(IJ,K11)=A(IJ,JJJ)
C(II,K22)=A(IJ,JJJ)
C(II,K21)=A(IJ,JJJ)
KKK=KKK+KL
217 IVK=IVK-KL
KZ=KK*KL+1
LZ=NCLS-(KK-1)*KL
218 IF (KZ-LZ) 220,218,218
DO 219 I=1,KL
DO 219 J=KZ,LZ
C(I,J)=0.0
219 LM1=NCLS-2*KL+1
LM3=NCLS-KL
DO 221 I=1,LM1,KL
LM2=I+KL-1
DO 221 II=I,LM2
III=II+KL
DO 221 JJ=II,LM3
JJJ=JJ+KL
221 C(III,JJJ)=C(II,JJ)
GO TO (222,224,222),LCOD

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MODEL _____

T6508000
T6508010
T6508020
T6508030
T6508040
T6508050
T6508060
T6508070
T6508080
T6508090
T6508100
T6508110
T6508120
T6508130
T6508140
T6508150
T6508160
T6508170
T6508180
T6508190
T6508200
T6508210
T6508220
T6508230
T6508240
T6508250
T6508260
T6508270
T6508280
T6508290
T6508300
T6508310
T6508320
T6508330
T6508340
T6508350
T6508360
T6508370
T6508380
T6508390

```

222 KKL=KL*(KK-1)
DO 223 III=1,NCLS,KL
JTST=KKL+1-III
JCLS=NCLS-KKL+III
JJJ=III+KL-1
JJ=III+KL*KK-1
DO 223 I=III,JJJ
DO 2220 J=I,JJ
2220 C(I,J)=P(I)*C(I,J)*P(J)
IF (JTST) 223,223,2221
2221 DO 2222 J=JCLS,NCLS
2222 C(I,J)=P(I)*C(I,J)*P(J)
223 CONTINUE
C
224 DO 2240 I=1,NCLS
DO 2240 J=I,NCLS
2240 C(J,I)=C(I,J)
C
DO 227 II=1,NRWS
SCL1=(1.0/(FM(II)**2*FW(II)**4))/((1.0-(FREQ/FF(II))**2)**2)+(2.0*(FREQ/FF(II))**2)**2
1FL(II)=(FREQ/FF(II))**2
R(II)=0.0
KKK=1
LLL=KL*(KK-1)
MMM=LLL
NNN=KL
ITST=NCLS+1-KL*KK
DO 2249 III=1,NCLS,KL
JJJ=III+KL-1
JCLS=NCLS+III-MMM
ITT=ITST-III
IMT=MMM-III+1
IF (IMT) 2244,2243,2241
2241 DO 2242 I=III,JJJ
DO 2242 J=JCLS,NCLS
2242 R(II)=R(II)+PHIA(II,J)*C(J,I)*PHIA(II,I)
2243 KKK=KKK-KL
2244 KKK=KKK+KL
LLL=LLL+KL
DO 2245 I=III,JJJ

```

SYMMETRIC MATRIX

CALC •-DIAGONAL MATRIX

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```

DO 2245 J=KKK,LLL
2245 R(II)=R(II)+PHIA(II,J)*C(J,I)*PHIA(II,I)
IF (ITT) 2246,2248,2249
2246 DO 2247 I=III,JJJ
DO 2247 J=1,NNN
2247 R(II)=R(II)+PHIA(II,J)*C(J,I)*PHIA(II,I)
NNN=NNN+KL
2248 LLL=LLL-KL
2249 CONTINUE
R(II)=SCL1*R(II)
227 CONTINUE
GO TO 2276
2270 IF(LCOD-2)12280,12270,12280
12270 DO 12271 I=1,KL
DO 12271 J=1,KL
12271 C(I,J) = A(I,J)
GO TO 12285
12280 DO 2271 I=1,KL
DO 2271 J=1,KL
2271 C(I,J)=P(I)*A(I,J)*P(J)
12285 LM1 = NCLS-2*KL+1
LM3=NCLS-KL
DO 2272 I=1,LM1,KL
LM2=I+KL-1
DO 2272 II=I,LM2
III=II+KL
DO 2272 JJ=II,LM3
JJJ=JJ+KL
2272 C(III,JJJ)=C(II,JJ)
3272 DO 2273 I=1,NCLS
DO 2273 J=1,NCLS
2273 C(J,I)=C(I,J)
DO 2275 II=1,NRWS
SCL1=(1.0/(FM(II)**2*FW(II)**4))/((1.0-(FREQ/FF(II))**2)**2) + (2.0*(FREQ/FF(II))**2)
1FL(II)=(FREQ/FF(II))**2)
R(II)=0.0
DO 2274 III=1,NCLS,KL
JJJ=III+KL-1
DO 2274 I=III,JJJ
DO 2274 J=III,JJJ

```

T6508400
T6508410
T6508420
T6508430
T6508440
T6508450
T6508460
T6508470
T6508480
T6508490
T6508500
T6508510
T6508520
T6508530
T6508540
T6508550
T6508560
T6508570
T6508580
T6508590
T6508600
T6508610
T6508620
T6508630
T6508640
T6508650
T6508660
T6508670
T6508680
T6508690
T6508700
T6508710
T6508720
T6508730
T6508740
T6508750
T6508760
T6508770
T6508780
T6508790

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MODEL _____

T6508800
T6508810
T6508820
T6508830
T6508840
T6508850
T6508860
T6508870
T6508880
T6508890
T6508900
T6508910
T6508920
T6508930
T6508940
T6508950
T6508960
T6508970
T6508980
T6508990
T6509000
T6509010
T6509020
T6509030
T6509040
T6509050
T6509060
T6509070
T6509080
T6509090
T6509100
T6509110
T6509120
T6509130
T6509140
T6509150
T6509160
T6509170
T6509180
T6509190

```

2274 R(II)=R(II)+PHIA(II,J)*C(J,I)*PHIA(II,I)
      R(II)=SCL1*R(II)
2275 CONTINUE
      C
2276 IF(NVAL)231,231,229
229  TDLF=DFRQ/2.0
      DO 230 I=1,MRWS
230  T(I)=T(I)+TDLF*(R(I)+R1(I))
231  OMG=(6.28318531*FREQ)**4
232  IF (NVAL) 235,235,233
233  TDLF=DFRQ/2.0
      DO 234 I=1,MRWS
234  TT(I)=TT(I)+TDLF*(OMG*R(I)+OMG1*R1(I))
      C
235  OMG1=OMG
      DO 236 I=1,MRWS
236  R1(I)=R(I)
      C
237  IF(KCNT-KPRC)245,237,245
      WRITE (6,1033)FREQ
      NLNS=2
238  WRITE (6,1034)
      NLNS=NLNS+2
      CALL DPRT (MRWS,R,NLNS)
      WRITE (6,1050)
      DO 2238 I=1,MRWS
2238  SC(I) = R(I)*OMG*ACCEL
      CALL DPRT (MRWS,SC,NLNS)
2410 WRITE (6,1047)
      NLNS=NLNS+2
      DO 2411 I=1,NPRS
      SC(I)=0.0
      DO 2411 J=1,MRWS
2411  SC(I) = SC(I) + PHI(J,I)*R(J)
      CALL DPRT (NPRS,SC,NLNS)
242  WRITE (6,1036)
      NLNS=NLNS+2
      CALL DTRM (MRWS,NPRS,R,PHI,AC)
      DO 243 I=1,NPRS
243  AC(I)=OMG*AC(I)*ACCEL

```

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MODEL _____

T6509200
T6509210
T6509220
T6509230
T6509240
T6509250
T6509260
T6509270
T6509280
T6509290
T6509300
T6509310
T6509320
T6509330
T6509340
T6509350
T6509360
T6509370
T6509380
T6509390
T6509400
T6509410
T6509420
T6509430
T6509440
T6509450
T6509460
T6509470
T6509480
T6509490
T6509500
T6509510
T6509520
T6509530
T6509540
T6509550
T6509560
T6509570
T6509580
T6509590

```

244 CALL DPRT (NPRS,AC,NLNS)
245 KCNT=0
246 KCNT=KCNT+1
247 IF (FREQ-FR(1,NIVL+1)+.0001) 2480,246,246
248 IF (NVAL-2) 247,990,247
249 NIVL=NIVL+1
250 DFRQ=FR(2,NIVL)
251 KPRC=FR(3,NIVL)+.0000001
252 IPS=1
253 IF (IMS-INS) 248,2481,2481
254 2481 GO TO(2482,2488,2498,2479),IPS
255 2479 IPS = 1
256 KCNT = KCNT1
257 GO TO 2482
258 2482 IF (FREQ+DFRQ-FFF(INS)+.0001) 248,2483,2483
259 2483 IF (FR(1,NIVL+1)-FFF(INS)+.0001) 248,2484,2484
260 2484 IF (ABS(FREQ+DFRQ-FFF(INS))-0.0001) 2486,2486,2485
261 2485 FREQ=FREQ
262 DFRQ=DFRQ
263 DFRQ=FFF(INS)-FREQ2
264 FREQ=FREQ+DFRQ
265 IPS=2
266 KCNT2=KCNT
267 KCNT=KPRC
268 GO TO 250
269 IPS=1
270 INS=INS+1
271 IF (KCNT-KPRC) 2487,248,248
272 2487 KCNT2=KCNT+1
273 KCNT=KPRC
274 IPS=4
275 INS=INS-1
276 GO TO 248
277 2488 IF (ABS(FREQ2+DFRQ2-FREQ)-0.0001) 2490,2490,2491
278 2490 IPS=1
279 INS=INS+1
280 KCNT=KCNT2
281 GO TO 2492
282 2491 DFRQ=FREQ2+DFRQ2-FREQ
283 IPS=3

```

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MODEL _____

T6509600
T6509610
T6509620
T6509630
T6509640
T6509650
T6509660
T6509670
T6509680
T6509690
T6509700
T6509710
T6509720
T6509730
T6509740
T6509750
T6509760
T6509770
T6509780
T6509790
T6509800
T6509810
T6509820
T6509830
T6509840
T6509850
T6509860
T6509870
T6509880
T6509890
T6509900
T6509910
T6509920
T6509930
T6509940
T6509950
T6509960
T6509970
T6509980
T6509990

```

2492 KCNT=KCNT2
2493 IF (IMS-INS-1) 248,2493,2493
2494 IF (FREQ+DFRQ-FFF(INS+1)+.0001) 2494,2495,2495
2495 IF (FREQ+DFRQ-FR(1,NIVL+1)+.0001) 248,248,2497
2496 IF (FFF(INS+1)-FR(1,NIVL+1)+.0001) 2496,2497,2497
2497 DFRQ = FFF(INS+1)-FREQ
2498 INS=INS+1
2499 IPS=2
2500 KCNT2=KCNT
2501 KCNT=KPRC
2502 GO TO 248
2503 INS = INS+1
2504 GO TO 248
2505 DFRQ = DFRQ2
2506 IPS=1
2507 INS=INS+1
2508 GO TO 2480
2509 FREQ1=FREQ
2510 FREQ=FREQ+DFRQ
2511 IF (FREQ-FR(1,NIVL+1)+.0001) 250,249,249
2512 FREQ=FR(1,NIVL+1)
2513 DFRQ=FREQ-FREQ1
2514 KPRC=KCNT
2515 IF (NIVLS-NIVL-1) 251,251,250
2516 NVAL=1
2517 GO TO 200
2518 NVAL=2
2519 GO TO 200
2520 WRITE(6,1037)
2521 NLNS=3
2522 CALL DPRT (MRWS,T,NLNS)
2523 WRITE (6,1038)
2524 NLNS=NLNS+3
2525 DO 9940 I=1,MRWS
2526 AC(I) = TT(I)*ACCEL
2527 CALL DPRT (MRWS,AC,NLNS)
2528 WRITE (6,1048)
2529 NLNS=NLNS+3
2530 DO 9971 I=1,NPRS
2531 SC(I)=0.0

```

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MODEL _____

```

DO 9971 J=1,MRWS
  SC(I) = SC(I)+PHI(J,I)*T(J)
  CALL DPRT (NPRS,SC,NLNS)
998  CALL DTRM (MRWS,NPRS,TT,PHI,AC)
    WRITE (6,1040)
    NLNS=NLNS+3
DO 9980 I=1,NPRS
  AC(I)=AC(I)*ACCEL
  CALL DPRT (NPRS,AC,NLNS)
999  CALL CLOCK(TM3,IM4,DT1,DT2)
    WRITE (6,1041)TM1,IM2,IM3,IM4,DT1,DT2
    RETURN
1032 FORMAT (1H0,10X,17HERROR, TLU ERR IN,13,7H TABLES)
1033 FORMAT (1H1,10X,12HFREQUENCY = ,1PE13.6/1H0,10X,9HDARC PRGM)
1034 FORMAT(1H0,10X36HNORMAL COORDINATE DEFLECTION SPECTRA)
1036 FORMAT(1H010X33HRADIAL SHELL ACCELERATION SPECTRA)
1037 FORMAT(1H010X40HMEAN-SQUARE NORMAL COORDINATE DEFLECTION)
1038 FORMAT(1H010X42HMEAN-SQUARE NORMAL COORDINATE ACCELERATION)
1040 FORMAT(1H010X31HMEAN-SQUARE RADIAL ACCELERATION)
1041 FORMAT (1H1,10X,13HINIT. TIME = ,A6,A2,4X,13HFINAL TIME = ,
1A6,A2,4X,7HDATE = ,A6,A2)
1047 FORMAT(1H010X31HRADIAL SHELL DEFLECTION SPECTRA)
1048 FORMAT(1H010X29HMEAN-SQUARE RADIAL DEFLECTION)
1050 FORMAT(1H0,10X38HNORMAL COORDINATE ACCELERATION SPECTRA)
END
END OF FILE CARD

```

T6510000
 T6510010
 T6510020
 T6510030
 T6510040
 T6510050
 T6510060
 T6510070
 T6510080
 T6510090
 T6510100
 T6510110
 T6510120
 T6510130
 T6510140
 T6510150
 T6510160
 T6510170
 T6510180
 T6510190
 T6510200
 T6510210
 T6510220
 T6510230
 T6510240
 T6510250

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4.9 Sample Case. - This section contains a sample case composed of a card input listing and the corresponding computer results. Only a portion of the output is given here as the entire listing is rather lengthy. A test case using all the options is impossible, but it is assumed that one case would aid immeasurably in understanding the program and its requirements.

4.9.1 Sample Input Listing. - The sample input listing, shown on the following pages, is an off-line listing of the cards as they would be setup for a computer run. To understand this section reference should be made to section 4.3

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MODEL _____

1111 DUP. RESPONSE NO. 10 CHECK CASE FOR CORRECTED RESPONSE CAL.

27 1

0.	100.	1.	0001
150.	5.	1.	0002
170.	2.	1.	0003
174.	5.	1.	0004
179.	5.	1.	0005
209.	2.	1.	0006
219.	5.	1.	0007
239.	5.	1.	0008
244.	2.	1.	0009
252.	5.	1.	0010
272.	2.	1.	0011
290.	5.	1.	0012
310.	3.	1.	0013
316.	5.	1.	0014
331.	2.	1.	0015
341.	3.	1.	0016
350.	5.	1.	0017
404.	2.	1.	0018
418.	5.	1.	0019
435.	3.	1.	0020
441.	5.	1.	0021
461.	3.	1.	0022
473.	5.	1.	0023
503.	2.	1.	0024
511.	5.	1.	0025
546.	3.	1.	0026
565.	1.	1.	0027
			0028
			0029
			0030
			0031
			0032
			0033
			0034
			0035
			0036
			0037
			0038
			0039
			0040

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MODEL _____

0041	1.	1110.086	1122.7795	1129.1262	1135.473	410.
0042	2.	80	1148.1665	1154.5132	1123.396	
0043	3.	20	1.			
0044	4.	4	126			
0045	5.	2	18.			
0046	1.	8	700.			
0047	1.	1116.4327	.00035	.000195	.000095	
0048	5.	1141.8197	.00005			
0049	6.	1011.				
0050	0.	4	4	12.7	11.	
0051	0.	6	2	32.	160.	
0052	0.					
0053	1110.086					
0054	20					
0055	4					
0056	2					
0057	8					
0058	1116.4327					
0059	1141.8197					
0060	1011.					
0061	.053					
0062	0.					
0063	320.					
0064	.0002					
0065	.00005					
0066	0.					
0067	320.					
0068	.0002					
0069	.00005					
0070	0.					
0071	320.					
0072	.0002					
0073	.00005					
0074	0.					
0075	325.					
0076	.000275					
0077	.00001					
0078						
0079						
0080						

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MODEL _____

MAC 291UM (REV 14 JUN 62)

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MODEL _____

0121
0122
0123
0124
0125
0126
0127
0128
0129
0130
0131
0132
0133
0134
0135
0136

277.
280.
288.
313.
336.
337.
348.
406.
415.
438.
464.
469.
507.
508.
529.
532.

.020
.020
.020
.020
.020
.020
.020
.020
.020
.020
.020
.020
.020
.020
.020
.020

79.74
66.61
109.24
66.85
80.57
77.69
66.02
67.07
78.29
64.96
77.67
80.29
67.56
64.31
77.95
73.77

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MODEL _____

4.9.2 Sample Output Listing. - The sample output listing, shown on the following pages, is a direct result from the sample input and the DARC program. It contains various printouts. Initially it contains the input and computed matrices. Intermediate printout consists of the normal coordinate response at the desired frequencies. The final portion consists of printouts of the various mean-square responses.

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MODEL _____

SHELL RESPONSE TO RANDOM PRESSURE EXCITATION

NAS 1-3170 06/12/65

111 DUP. RESPONSE NO. 10 CHECK CASE FOR CORRECTED RESPONSE CAL.

NO. OF CALC. FREQ. = 27

CALC FREQ	CALC INTERVAL	NO. CALC / NO. PRINTOUT
0.	1.000000E 02	1.000000E 00
1.500000E 02	5.000000E 00	1.000000E 00
1.700000E 02	2.000000E 00	1.000000E 00
1.740000E 02	5.000000E 00	1.000000E 00
1.790000E 02	5.000000E 00	1.000000E 00
2.000000E 02	2.000000E 00	1.000000E 00
2.100000E 02	5.000000E 00	1.000000E 00
2.300000E 02	5.000000E 00	1.000000E 00
2.440000E 02	2.000000E 00	1.000000E 00
2.520000E 02	5.000000E 00	1.000000E 00
2.720000E 02	2.000000E 00	1.000000E 00
2.900000E 02	5.000000E 00	1.000000E 00
3.100000E 02	3.000000E 00	1.000000E 00
3.150000E 02	5.000000E 00	1.000000E 00
3.310000E 02	2.000000E 00	1.000000E 00
3.410000E 02	3.000000E 00	1.000000E 00
3.500000E 02	5.000000E 00	1.000000E 00
4.040000E 02	2.000000E 00	1.000000E 00
4.100000E 02	5.000000E 00	1.000000E 00
4.250000E 02	3.000000E 00	1.000000E 00
4.410000E 02	5.000000E 00	1.000000E 00
4.610000E 02	3.000000E 00	1.000000E 00
4.730000E 02	5.000000E 00	1.000000E 00
5.020000E 02	2.000000E 00	1.000000E 00
5.110000E 02	5.000000E 00	1.000000E 00
5.460000E 02	5.000000E 00	1.000000E 00
5.650000E 02	3.000000E 00	1.000000E 00
	-0.	-0.

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MODEL _____

PHIAT CALC CODE = 2 PHI2T CALC CODE = 2
ACCEL. SCALAR = 6.7000E-04

ALPHA = 8.820000E 01

MODES = 10

N	M	AMN
2.0000E 00	1.0000E 00	1.000000E 00
3.0000E 00	1.0000E 00	1.000000E 00
3.0000E 00	2.0000E 00	1.000000E 00
2.0000E 00	2.0000E 00	1.000000E 00
3.0000E 00	3.0000E 00	1.000000E 00
1.0000E 00	1.0000E 00	1.000000E 00
4.0000E 00	3.0000E 00	1.000000E 00
4.0000E 00	1.0000E 00	1.000000E 00
4.0000E 00	2.0000E 00	1.000000E 00
2.0000E 00	3.0000E 00	1.000000E 00
5.0000E 00	3.0000E 00	1.000000E 00
1.0000E 00	2.0000E 00	1.000000E 00
1.0000E 00	3.0000E 00	1.000000E 00
5.0000E 00	2.0000E 00	1.000000E 00
5.0000E 00	1.0000E 00	1.000000E 00
6.0000E 00	3.0000E 00	1.000000E 00
0.	3.0000E 00	1.000000E 00
0.	1.0000E 00	1.000000E 00
0.	2.0000E 00	1.000000E 00

X1 = 1.110084E 03 XF = 1.160860E 03

PHIAT CODE = 2 N COLS = 80 NT = 20 NX = 4

PHI2T CODE = 2 N COLS = 8

PHI2T RESPONSE POINTS

1.11643E 03	1.12278E 03	1.12913E 03	1.13547E 03
1.14182E 03	1.14817E 03	1.15451E 03	1.12340E 03

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MODEL

SHELL RESPONSE TO RANDOM PRESSURE EXCITATION
NAS 1-3179 06/12/65

CODE = 2 N = 19 M = 80 PHAT

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MODEL

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-1.6097343F 03	6.8865886E 03	6.8789739E 02	-1.7145012E 03	-2.5072200E 03	1.0726115E 03	1.0714254E 03	-2.6703985E 03
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-1.9937239F 03	-2.0267533E 03	2.0852771E 03	2.1190639E 03	1.9937237E 03	2.0267531E 03	-2.0852769E 03	-2.1190637E 03
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1.9937240F 03	2.0267535E 03	-2.0852773E 03	-2.1190641E 03	-1.9937235E 03	-2.0267529E 03	2.0852767E 03	2.1190635E 03
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CODE = 2 N = 19 M = 80 PH1AT

27

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MODEL

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CODE = 2 N = 10 M = 8 PHI2T

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1.8009200E 02	6.2189218E 02	1.0736717E 03	1.2720694E 03	1.0979545E 03	6.5033562E 02	1.9259321E 02	6.7002733E 02
6.1487430E 02	1.2437873E 03	6.2895342E 02	1.0857614E-09	6.4315478E 02	1.3006694E 03	6.5755261E 02	1.2379206E 03
6.1487430E 02	1.2437873E 03	6.2895342E 02	1.0857614E-09	6.4315478E 02	1.3006694E 03	6.5755261E 02	1.2379206E 03
1.0496615E 03	6.2189801E 02	1.8420379E 02	1.2720694E 03	1.8838916E 02	6.5033202E 02	1.1225071E 03	4.8132775E 02
1.8009200E 02	6.2189218E 02	1.0736717E 03	1.2720694E 03	1.0979545E 03	6.5033562E 02	1.9259321E 02	6.7002733E 02
1.0496615E 03	6.2189801E 02	1.8420379E 02	1.2720694E 03	1.8838916E 02	6.5033202E 02	1.1225071E 03	4.8132775E 02
1.8009200E 02	6.2189218E 02	1.0736717E 03	1.2720694E 03	1.0979545E 03	6.5033562E 02	1.9259321E 02	6.7002733E 02
6.1487430E 02	1.2437873E 03	6.2895342E 02	1.0857614E-09	6.4315478E 02	1.3006694E 03	6.5755261E 02	1.2379206E 03
1.0496615E 03	6.2189801E 02	1.8420379E 02	1.2720694E 03	1.8838916E 02	6.5033202E 02	1.1225071E 03	4.8132775E 02
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6.1487430E 02	1.2437873E 03	6.2895342E 02	1.0857614E-09	6.4315478E 02	1.3006694E 03	6.5755261E 02	1.2379206E 03
1.0496615E 03	6.2189801E 02	1.8420379E 02	1.2720694E 03	1.8838916E 02	6.5033202E 02	1.1225071E 03	4.8132775E 02
6.1487430E 02	1.2437873E 03	6.2895342E 02	1.0857614E-09	6.4315478E 02	1.3006694E 03	6.5755261E 02	1.2379206E 03
1.8009200E 02	6.2189218E 02	1.0736717E 03	1.2720694E 03	1.0979545E 03	6.5033562E 02	1.9259321E 02	6.7002733E 02
1.0496615E 03	6.2189801E 02	1.8420379E 02	1.2720694E 03	1.8838916E 02	6.5033202E 02	1.1225071E 03	4.8132775E 02
1.8009200E 02	6.2189218E 02	1.0736717E 03	1.2720694E 03	1.0979545E 03	6.5033562E 02	1.9259321E 02	6.7002733E 02
6.1487430E 02	1.2437873E 03	6.2895342E 02	1.0857614E-09	6.4315478E 02	1.3006694E 03	6.5755261E 02	1.2379206E 03
1.0496615E 03	6.2189801E 02	1.8420379E 02	1.2720694E 03	1.8838916E 02	6.5033202E 02	1.1225071E 03	4.8132775E 02
1.8009200E 02	6.2189218E 02	1.0736717E 03	1.2720694E 03	1.0979545E 03	6.5033562E 02	1.9259321E 02	6.7002733E 02
6.1487430E 02	1.2437873E 03	6.2895342E 02	1.0857614E-09	6.4315478E 02	1.3006694E 03	6.5755261E 02	1.2379206E 03

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SHELL RESPONSE TO RANDOM PRESSURE EXCITATION

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PSD CODE = 1 CO-SPEC. CODE = 0 FREQ. MULT. = 1.000000E 00 PSD MULT. = 1.000000E 00

KL = 4 KC = 4 KK = 2

EMPIRICAL CO-SPECTRA METHOD

LONGITUDINAL DECAY RATE 0.52999999E-01
 CIRCUMFERENTIAL DECAY RATE 0.12599999E-00
 AVERAGE SIZE OF AREA (LONG.) 0.12700000E 02
 AVERAGE SIZE OF AREA (CIR.) 0.11000000E 02
 CONVECTION VELOCITY 0.41000000E 03

PSD TABLE, ELE (1, 1) WITH 6 PTS.

FREQ.

0.	1.800E 01	3.200E 01	1.600E 02	3.200E 02	7.000E 02
PSD					
2.000E-04	3.500E-04	1.950E-04	9.500E-05	5.000E-05	5.000E-05

PSD TABLE, ELE (2, 2) WITH 6 PTS.

FREQ.

0.	1.800E 01	3.200E 01	1.600E 02	3.200E 02	7.000E 02
PSD					
2.000E-04	3.500E-04	1.950E-04	9.500E-05	5.000E-05	5.000E-05

PSD TABLE, ELE (3, 3) WITH 6 PTS.

FREQ.

0.	1.800E 01	3.200E 01	1.600E 02	3.200E 02	7.000E 02
PSD					
2.000E-04	3.500E-04	1.950E-04	9.500E-05	5.000E-05	5.000E-05

PSD TABLE, ELE (4, 4) WITH 6 PTS.

FREQ.

0.	1.950E 01	5.200E 01	1.600E 02	3.250E 02	7.000E 02
PSD					
2.750E-04	5.250E-04	9.000E-05	4.500E-05	1.000E-05	10.000E-06

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SHELL RESPONSE TO RANDOM PRESSURE EXCITATION NAS 1-3179 06/12/65

CO- TBLF, ELF (1,1) TO BE CALCULATED

CO- TBLF, ELF (2,1) WITH 19 PTS.

FRFQ.	2.000E 01	7.200E 01	9.750E 01	1.350E 02	1.500E 02	1.850E 02	2.100E 02	2.400E 02	2.700E 02
0.	2.000E 01	7.200E 01	9.750E 01	1.350E 02	1.500E 02	1.850E 02	2.100E 02	2.400E 02	2.700E 02
2.970E 02	3.250E 02	3.500E 02	3.900E 02	4.100E 02	4.300E 02	4.600E 02	5.000E 02	7.000E 02	
CO-SPEC.									
6.700E-01	6.500E-01	0.	-2.500E-01	-5.200E-01	-5.200E-01	-2.500E-01	0.	2.200E-01	3.600E-01
3.600E-01	2.200E-01	0.	-2.000E-01	-2.600E-01	-2.600E-01	-2.000E-01	0.	0.	

CO- TBLF, ELF (3,1) WITH 19 PTS.

FRFQ.	2.000E 01	5.500E 01	7.000E 01	7.750E 01	1.120E 02	1.350E 02	1.400E 02	1.630E 02	1.900E 02
0.	2.000E 01	5.500E 01	7.000E 01	7.750E 01	1.120E 02	1.350E 02	1.400E 02	1.630E 02	1.900E 02
2.050E 02	2.150E 02	2.500E 02	2.750E 02	2.850E 02	3.350E 02	3.600E 02	3.800E 02	7.000E 02	
CO-SPEC.									
5.400E-01	3.000E-01	-3.000E-01	-4.200E-01	-4.200E-01	1.200E-01	3.200E-01	3.200E-01	1.200E-01	-1.200E-01
-2.100E-01	-2.100E-01	4.000E-02	1.600E-01	1.600E-01	-1.200E-01	-1.200E-01	0.	0.	

CO- TBLF, ELF (4,1) TO BE CALCULATED

CO- TBLF, ELF (1,2) WITH 13 PTS.

FRFQ.	1.902E 01	2.853E 01	5.706E 01	7.608E 01	9.510E 01	1.522E 02	1.712E 02	1.997E 02	2.568E 02
0.	1.902E 01	2.853E 01	5.706E 01	7.608E 01	9.510E 01	1.522E 02	1.712E 02	1.997E 02	2.568E 02
3.320E 02	4.500E 02	7.000E 02							
CO-SPEC.									
5.800E-01	3.800E-01	3.400E-01	3.700E-01	4.100E-01	2.500E-01	3.200E-01	1.900E-01	2.000E-01	1.000E-01
1.000E-01	0.	0.							

CO- TBLF, ELF (2,2) TO BE CALCULATED

CO- TBLF, ELF (3,2) TO BE CALCULATED

CO- TBLF, ELF (4,2) TO BE CALCULATED

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SHELL RESPONSE TO RANDOM PRESSURE EXCITATION NAS 1-3179 06/12/65

NO MODES = 19

GFN. MASSES	DMP. COEFFS	MODAL FREQS
9.034000E 01	2.000000E-02	1.720000E 02
8.308000E 01	2.000000E-02	2.150000E 02
7.860000E 01	2.000000E-02	2.480000E 02
7.974000E 01	2.000000E-02	2.770000E 02
6.661000E 01	2.000000E-02	2.800000E 02
1.092400E 02	2.000000E-02	2.880000E 02
6.685000E 01	2.000000E-02	3.130000E 02
8.057000E 01	2.000000E-02	3.360000E 02
7.759000E 01	2.000000E-02	3.370000E 02
6.602000E 01	2.000000E-02	3.480000E 02
6.707000E 01	2.000000E-02	4.060000E 02
7.829000E 01	2.000000E-02	4.150000E 02
6.496000E 01	2.000000E-02	4.380000E 02
7.767000E 01	2.000000E-02	4.640000E 02
8.029000E 01	2.000000E-02	4.690000E 02
6.756000E 01	2.000000E-02	5.070000E 02
6.431000E 01	2.000000E-02	5.080000E 02
7.795000E 01	2.000000E-02	5.290000E 02
7.377000E 01	2.000000E-02	5.320000E 02

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FREQUENCY = 0.

NARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	3	4	5	6	7	8
4.0767530E-11	1.7114507E-11	4.0425515E-12	3.0947398E-12	1.9424110E-12	3.8609621E-12	8.3703386E-13	2.4898702E-12	12
9	10	11	12	13	14	15	16	16
8.8909585E-13	1.0566310E-12	1.5407961E-13	7.1607644E-13	4.9804732E-13	1.5681598E-13	5.0643803E-13	1.3296963E-14	14
17	18	19						
5.8676862E-13	1.3702965E-12	6.2045041E-13						

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	3	4	5	6	7	8
0.	0.	0.	0.	0.	0.	0.	0.	0.
9	10	11	12	13	14	15	16	16
0.	0.	0.	0.	0.	0.	0.	0.	0.
17	18	19						
0.	0.	0.						

RADIAL SHELL DEFLECTION SPECTRA

1	1	2	3	4	5	6	7	8
2.3100758E-08	5.6119063E-08	7.7905315E-08	9.0568628E-08	7.9667138E-08	5.8685681E-08	2.4704191E-08	5.8530145E-08	8
1	2	3	4	5	6	7	8	8

RADIAL SHELL ACCELERATION SPECTRA

1	1	2	3	4	5	6	7	8
0.	0.	0.	0.	0.	0.	0.	0.	0.
17	18	19						
0.	0.	0.						

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FREQUENCY = 1.000000E 02

DARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
7.7117968E-12	2.3728923E-12	1.7614863E-12	1.1633486E-12	1.2772203E-12	4.0823691E-13	6.3963705E-13	2.6593522E-13								
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
3.7046657E-13	5.6C71584E-13	1.6257511E-13	2.2095028E-13	2.3356171E-13	7.4298844E-14	5.2365173E-14	4.7564278E-14								
17	17	18	18	19	19										
2.6278124E-13	1.1957624E-13	1.8038091E-13													

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
8.0528539E-06	2.4778344E-06	1.8393887E-06	1.2147981E-06	1.3337058E-06	4.2629136E-07	6.6792525E-07	2.7769631E-07								
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
3.8685060E-07	5.8551372E-07	1.6976506E-07	2.3072189E-07	2.4389107E-07	7.7584739E-08	5.4681043E-08	4.9667826E-08								
17	17	18	18	19	19										
2.7440285E-07	1.2528224E-07	1.8835832E-07													

RADIAL SHELL DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
7.6294509E-09	1.3468423E-08	1.4694778E-08	1.7955587E-08	1.5027096E-08	1.4084378E-08	8.1589856E-09	1.3524891E-08								

RADIAL SHELL ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
7.0669663E-03	1.4064069E-02	1.5344660E-02	1.8749680E-02	1.5691675E-02	1.4707265E-02	8.5198200E-03	1.4123035E-02								

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FREQUENCY = 1.500000E 02

QARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
2.0905711E-11	2.7189114E-12	1.2910120E-12	7.6466256E-13	1.1118700E-12	3.1593123E-13	4.9702096E-13	1.5679198E-13								
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
1.0134656E-13	4.1848382E-13	1.0732771E-13	1.1461640E-13	1.5927377E-13	3.0994551E-14	2.3842809E-14	2.8391750E-14								
17	17	18	18	19	19										
1.7267416E-13	7.0735962E-14	8.8026892E-14													

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.5899329E-04	1.4373229E-05	6.8247944E-06	4.0423054E-06	5.8777796E-06	1.6701361E-06	2.6274472E-06	8.2886370E-07								
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
1.0115328E-06	2.2122691E-06	5.6737626E-07	6.0590711E-07	8.4198345E-07	1.6384932E-07	1.2604242E-07	1.5008989E-07								
17	17	18	18	19	19										
9.1282313E-07	3.7393796E-07	4.6534458E-07													

RADIAL SHELL DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.0121842E-08	2.5278870E-08	3.7657042E-08	4.5396295E-08	3.8508703E-08	2.6435012E-08	1.0824405E-08	2.6511286E-08								

RADIAL SHELL ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
5.3508016E-02	1.3363400E-01	1.9906985E-01	2.3998257E-01	2.0357206E-01	1.3974581E-01	5.7222036E-02	1.4014903E-01								

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FREQUENCY = 1.550000CE 02

DARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	2	3	4	4	5	5	6	6	7	7	8	8
4.4658449E-11	2.9368074E-12	1.3690879E-12	7.8054793E-13	1.0676674E-12	3.0815109E-13	4.7360930E-13	1.5705481E-13							
9	9	10	10	11	12	12	13	13	14	14	15	15	16	16
1.9891999E-13	3.9062125E-13	1.0135554E-13	1.1187158E-13	1.4639704E-13	3.3021192E-14	2.4350967E-14	2.6990141E-14							
17	17	18	18	19										
1.5778920E-13	6.6195583E-14	8.4945576E-14												

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	2	3	4	4	5	5	6	6	7	7	8	8
2.6916878E-04	1.7700947E-05	8.2518698E-06	4.7045774E-06	6.4351253E-06	1.8573115E-06	2.8545737E-06	9.4661261E-07							
9	9	10	10	11	12	12	13	13	14	14	15	15	16	16
1.1989456E-06	2.3543819E-06	6.1089776E-07	6.7428082E-07	8.8237530E-07	1.9902782E-07	1.4676999E-07	1.6267701E-07							
17	17	18	18	19										
9.5103899E-07	3.9857901E-07	5.1201449E-07												

RADIAL SHELL DEFLECTION SPECTRA

1	1	2	2	3	4	4	5	5	6	6	7	7	8	8
1.2738852E-08	3.4622146E-08	5.3755603E-08	6.4259140E-08	5.4971354E-08	3.6205620E-08	1.3623078E-08	3.6592406E-08							

RADIAL SHELL ACCELERATION SPECTRA

1	1	2	2	3	4	4	5	5	6	6	7	7	8	8
7.6780571E-02	2.0867721E-01	3.2399983E-01	3.8730753E-01	3.3132749E-01	2.1822125E-01	8.2110044E-02	2.2055251E-01							

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FREQUENCY = 1.600000E 02

NARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
7.4391772E-11	3.0883953E-12	1.4204202E-12	7.7411148E-13	1.0093665E-12	2.8531309E-13	4.4639291E-13	1.5333672E-13	9	9	10	10	11	11	12	12
2.0263955E-13	3.5691949E-13	9.5355766E-14	1.0551428E-13	1.3133743E-13	3.4645078E-14	2.4653280E-14	2.5845731E-14	17	17	18	18	19	19	20	20
1.4059990E-13	5.8552229E-14	7.9112923E-14													

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
5.0902687E-04	2.1135235E-05	9.7205544E-06	5.2975822E-06	6.9075345E-06	1.9525218E-06	3.0548612E-06	1.0493500E-06	9	9	10	10	11	11	12	12
1.3867507E-06	2.4425556E-06	6.5256105E-07	7.2208018E-07	8.9879925E-07	2.3709137E-07	1.6871313E-07	1.7687360E-07	17	17	18	18	19	19	20	20
9.6218636E-07	4.0069841E-07	5.4140419E-07													

RADIAL SHELL DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.7969883E-08	5.3128718E-08	8.5789763E-08	1.0201721E-07	8.7730011E-08	5.5558628E-08	1.9217228E-08	5.6557453E-08								

RADIAL SHELL ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.2297574E-01	3.6358297E-01	5.8709673E-01	6.9814823E-01	6.0037468E-01	3.8021190E-01	1.3151186E-01	3.8704729E-01								

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FREQUENCY = 1.650000E 02

DAPC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	2	3	4	5	6	7	8
1.6811026E-10	3.3768046E-12	1.5182802E-12	7.8939855E-13	9.7810689E-13	2.7021351E-13	4.3102520E-13	1.5339757E-13
9	10	11	12	13	14	15	16
2.1103647E-13	3.3333764E-13	9.18317C9E-14	1.0203647E-13	1.2027880E-13	3.6842025E-14	2.5431818E-14	2.5313579E-14
17	18	19					
1.2783673E-13	5.2800898E-14	7.5506732E-14					

NORMAL COORDINATE ACCELERATION SPECTRA

1	2	3	4	5	6	7	8
1.3011407E-03	2.6135810E-05	1.1751193E-05	6.1097911E-06	7.5703569E-06	2.0914000E-06	3.3360511E-06	1.1872673E-06
9	10	11	12	13	14	15	16
1.6333812E-06	2.5799684E-06	7.1075954E-07	7.8974243E-07	9.3093445E-07	2.8515009E-07	1.9683732E-07	1.9592217E-07
17	18	19					
9.8943141E-07	4.0866867E-07	5.8440741E-07					

RADIAL SHELL DEFLECTION SPECTRA

1	2	3	4	5	6	7	8
3.4865740E-08	1.1166874E-07	1.8676686E-07	2.2146300E-07	1.9099085E-07	1.1677609E-07	3.7286993E-08	1.1963513E-07

RADIAL SHELL ACCELERATION SPECTRA

1	2	3	4	5	6	7	8
2.6986178E-01	8.6429430E-01	1.4455391E 00	1.7140805E 00	1.4782320E 00	9.0382419E-01	2.8859406E-01	9.2595261E-01

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FREQUENCY = 1.700000E 02

NARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
5.6610497E-10	3.7542095E-12	1.6305016E-12	8.0686338E-13	9.4927363E-13	2.5508691E-13	4.1654094E-13	1.5328633E-13	1.5328633E-13	1.5328633E-13	1.5328633E-13	1.5328633E-13	1.5328633E-13	1.5328633E-13	1.5328633E-13	1.5328633E-13
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
2.1901693E-13	3.1092822E-13	8.8428489E-14	9.8598453E-14	1.0982780E-13	3.8761191E-14	2.6073402E-14	2.4765931E-14	2.4765931E-14	2.4765931E-14	2.4765931E-14	2.4765931E-14	2.4765931E-14	2.4765931E-14	2.4765931E-14	2.4765931E-14
17	17	18	18	19	19										
1.1583868E-13	4.7286761E-14	7.1976967E-14													

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
4.9372703E-03	3.2742244E-05	1.4220903E-05	7.0370389E-06	8.2790662E-06	2.2247341E-06	3.6328514E-06	1.3368829E-06	1.3368829E-06	1.3368829E-06	1.3368829E-06	1.3368829E-06	1.3368829E-06	1.3368829E-06	1.3368829E-06	1.3368829E-06
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
1.9101507E-06	2.7109805E-06	7.7122685E-07	8.5992392E-07	9.5786039E-07	3.3805475E-07	2.2739852E-07	2.1599545E-07	2.1599545E-07	2.1599545E-07	2.1599545E-07	2.1599545E-07	2.1599545E-07	2.1599545E-07	2.1599545E-07	2.1599545E-07
17	17	18	18	19	19										
1.0102841E-06	4.1241030E-07	6.2774531E-07													

RADIAL SHELL DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.0659152E-07	3.5950839E-07	6.1453258E-07	7.2807704E-07	6.2843111E-07	3.7595116E-07	1.1399060E-07	3.8666170E-07	3.8666170E-07	3.8666170E-07	3.8666170E-07	3.8666170E-07	3.8666170E-07	3.8666170E-07	3.8666170E-07	3.8666170E-07
17	17	18	18	19	19										

RADIAL SHELL ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
9.2963526E-01	3.1354434E 00	5.3596305E 00	6.3499057E 00	5.4808462E 00	3.2788486E 00	9.9416614E-01	3.3722603E 00	3.3722603E 00	3.3722603E 00	3.3722603E 00	3.3722603E 00	3.3722603E 00	3.3722603E 00	3.3722603E 00	3.3722603E 00
17	17	18	18	19	19										

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FREQUENCY = 1.720000F 02

DARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
7.2346091E-10	3.9002278E-12	1.6916337E-12	8.2186900E-13	9.4329169E-13	2.5403251E-13	4.1222804E-13	1.5430519E-13								
9	10	10	11	11	13	13	14	14	15	15	16	16	16	16	16
2.2289382E-13	3.0412934E-13	8.7107933E-14	9.8241210E-14	1.0660954E-13	3.9438998E-14	2.6299098E-14	2.4455535E-14								
17	17	18	18	19	19										
1.1211673E-13	4.6098267E-14	7.1349424E-14													

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
6.6119342E-03	3.6467478E-05	1.5460174E-05	7.5112229E-06	8.6209288E-06	2.3216532E-06	3.7674334E-06	1.4102256E-06								
9	10	10	11	11	13	13	14	14	15	15	16	16	16	16	16
2.0370704E-06	2.7794980E-06	7.9609657E-07	8.9784579E-07	9.7432556E-07	3.6044078E-07	2.4035264E-07	2.2350396E-07								
17	17	18	18	19	19										
1.0246569E-06	4.2130115E-07	6.5207645E-07													

RADIAL SHELL DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.3499614E-07	4.5760173E-07	7.8378856E-07	9.2852260E-07	8.0151509E-07	4.7853097E-07	1.4436694E-07	4.9234420E-07								

RADIAL SHELL ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.2337563F 00	4.1821124E 00	7.1631982E 00	8.4859512E 00	7.3252046E 00	4.3733889E 00	1.3193980E 00	4.4996308E 00								

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FRQUENCY = 1.740000E 02

DARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	3	4	5	6	7	8
5.2949566E-10	4.3384491E-12	1.7740617E-12	8.4863423E-13	9.4503936E-13	2.6049194E-13	4.0988171E-13	1.5690733E-13	
9	10	11	12	13	14	15	16	
2.2793729E-13	3.0053558E-13	8.5816922E-14	9.9372641E-14	1.0462971E-13	4.0064208E-14	2.6507674E-14	2.4018110E-14	
17	18	19						
1.0975371E-13	4.4307845E-14	7.1845396E-14						

NORMAL COORDINATE ACCELERATION SPECTRA

1	2	3	4	5	6	7	8
5.0481987E-03	4.1526539E-05	1.6980870E-05	8.1229125E-06	9.0456781E-06	2.4933631E-06	3.9232843E-06	1.5018774E-06
9	10	11	12	13	14	15	16
2.1817582E-06	2.8766507E-06	8.2141792E-07	9.5116981E-07	1.0014891E-06	3.8348448E-07	2.5372476E-07	2.2989528E-07
17	18	19					
1.0505348E-06	4.4324700E-07	6.8768598E-07					

RADIAL SHELL DEFLECTION SPECTRA

1	2	3	4	5	6	7	8
1.0018821E-07	3.3733162E-07	5.7597933E-07	6.8221586E-07	5.8900594E-07	3.5276009E-07	1.0714280E-07	3.6275516E-07

RADIAL SHELL ACCELERATION SPECTRA

1	2	3	4	5	6	7	8
0.5897623E-01	3.2288531E 00	5.5131287E 00	6.5299979E 00	5.6378161F 00	3.3765305E 00	1.0255438E 00	3.4722008E 00

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FREQUENCY = 1.790000E 02

DARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.3334315E-10	5.4457036E-12	2.0004959E-12	9.1805257E-13	9.5306261E-13	2.7725522E-13	4.0500240E-13	1.6329350E-13								
9	10	10	11	11	13	13	14	14	15	15	16	16	16	16	16
2.3990360E-13	2.9229755E-13	8.2740401E-14	1.0185006E-13	9.9969187E-14	4.1393766E-14	2.6956426E-14	2.2961437E-14								
17	17	18	18	19	19										
1.0417774E-13	4.6698394E-14	7.2788986E-14													

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.4294767E-03	5.8379499E-05	2.1445888E-05	9.8417859E-06	1.0217104E-05	2.9722552E-06	4.3417414E-06	1.7505530E-06								
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
2.5718352E-06	3.1335133E-06	8.8700075E-07	1.0918618E-06	1.0716982E-06	4.4375300E-07	2.8898059E-07	2.4615317E-07								
17	17	18	18	19	19										
1.1168151E-06	5.0062012E-07	7.8031873E-07													

RADIAL SHELL DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
2.9719806E-08	9.2047666E-08	1.5204921E-07	1.7969667E-07	1.5548799E-07	9.6257606E-08	3.1248072E-08	9.8457694E-08								

RADIAL SHELL ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
3.1324467E-01	9.8677728E-01	1.6300110E 00	1.9263996E 00	1.6668757E 00	1.0319090E 00	3.3498827E-01	1.0554946E 00								

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FREQUENCY = 1.840000E 02

JARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	2	3	4	5	6	7	8
5.0247141E-11	7.0658571E-12	2.2626926E-12	9.9157754E-13	9.6734629E-13	2.9506712E-13	4.0177754E-13	1.6953223E-13
9	10	11	12	13	14	15	16
2.5090616E-13	2.8526852E-13	7.9909318E-14	1.0380782E-13	9.5757954E-14	4.2387904E-14	2.7304621E-14	2.1964658E-14
17	18	19					
9.9108641E-14	4.6004244E-14	7.3303778E-14					

NORMAL COORDINATE ACCELERATION SPECTRA

1	2	3	4	5	6	7	8
6.0141869E-04	8.4572741E-05	2.7082647E-05	1.1868402E-05	1.1578373E-05	3.5317209E-06	4.8089605E-06	2.0291672E-06
9	10	11	12	13	14	15	16
3.0031400E-06	3.4144394E-06	9.5645154E-07	1.2424978E-06	1.1461472E-06	5.0734980E-07	3.2681480E-07	2.6289965E-07
17	18	19					
1.1842523E-06	5.6140683E-07	8.7738844E-07					

RADIAL SHELL DEFLECTION SPECTRA

1	2	3	4	5	6	7	8
1.4756883E-08	4.1823692E-08	6.4815896E-08	7.6073482E-08	6.6281755E-08	4.3736542E-08	1.5781195E-08	4.4312274E-08

RADIAL SHELL ACCELERATION SPECTRA

1	2	3	4	5	6	7	8
1.7642876E-01	5.0059663E-01	7.7579519E-01	9.1053963E-01	7.9334037E-01	5.2349194E-01	1.8888846E-01	5.3038300E-01

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DARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
2.5518405E-11	9.8038853E-12	2.5797024E-12	1.0731815E-12	9.8527850E-13	3.2062784E-13	3.9871927E-13	3.9871927E-13	1.7957260E-13							
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
2.6172110E-13	2.7851740E-13	7.6971654E-14	1.0552846E-14	9.1682628E-14	4.3191183E-14	2.8197978E-14	2.8197978E-14	2.0918756E-14							
17	17	18	18	19	19										
9.4221982E-14	4.7885879E-14	7.3595175E-14													

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
3.4001260E-04	1.3062903E-04	3.4372497E-05	1.4299295E-05	1.3128058E-05	4.2721128E-06	5.3126193E-06	5.3126193E-06	2.3926631E-06							
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
3.4872269E-06	3.7110243E-06	1.0255865E-06	1.4060833E-06	1.2215986E-06	5.7548839E-07	3.7571578E-07	3.7571578E-07	2.7872590E-07							
17	17	18	18	19	19										
1.2554335E-06	6.3809481E-07	9.8059756E-07													

RADIAL SHELL DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.1051698E-08	2.8680898E-08	4.1503733E-08	4.8141322E-08	4.2442344E-08	2.9992635E-08	1.1818811E-08	1.1818811E-08	3.0111017E-08							

RADIAL SHELL ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.4725515E-01	3.8215031E-01	5.5300446E-01	6.4144509E-01	5.6551071E-01	3.9962817E-01	1.5747632E-01	1.5747632E-01	4.0120552E-01							

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FREQUENCY = 1.940000E 02

DARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	2	3	4	5	6	7	8
1.5208200E-11	1.4706353E-11	2.9376990E-12	1.1520335E-12	1.0042076E-12	3.5299910E-13	3.9442310E-13	1.9233422E-13
9	10	11	12	13	14	15	16
2.6016844E-13	2.7120025E-13	7.3635075E-14	1.0586565E-13	8.7477674E-14	4.3263616E-14	2.9407706E-14	1.9739180E-14
17	18	19					
8.9249254E-14	4.9303315E-14	7.2883031E-14					

NORMAL COORDINATE ACCELERATION SPECTRA

1	2	3	4	5	6	7	8
2.2627771E-04	2.1752349E-04	4.3451869E-05	1.7039870E-05	1.4853360E-05	5.2212387E-06	5.8339610E-06	2.8448392E-06
9	10	11	12	13	14	15	16
3.0813038E-06	4.0113566E-06	1.0891455E-06	1.5658719E-06	1.2938931E-06	6.3991750E-07	4.3497301E-07	2.9196436E-07
17	18	19					
1.3270668E-06	7.2925143E-07	1.0780219E-06					

RADIAL SHELL DEFLECTION SPECTRA

1	2	3	4	5	6	7	8
1.0368924E-08	2.5951910E-08	3.6123589E-08	4.1429437E-08	3.6940509E-08	2.7138725E-08	1.1088640E-08	2.7126551E-08

RADIAL SHELL ACCELERATION SPECTRA

1	2	3	4	5	6	7	8
1.5336804E-01	3.8335644E-01	5.3430848E-01	6.1278794E-01	5.4639165E-01	4.0141224E-01	1.6401344E-01	4.0123217E-01

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FREQUENCY = 1.990000E 02

QAPC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	3	4	5	6	7	8
1.0057929E-11	2.4300962E-11	3.3748009E-12	1.2360959E-12	1.0312102E-12	3.8861598E-13	3.9163469E-13	2.0526081E-13	
9	10	11	12	13	14	15	16	
2.7509895E-13	2.4497989E-13	7.0481279E-14	1.0551979E-13	8.3644599E-14	4.2954588E-14	3.0494963E-14	1.8606968E-14	
17	18	19						
8.44696178E-14	5.0489560E-14	7.1674380E-14						

NORMAL COORDINATE ACCELERATION SPECTRA

1	2	3	4	5	6	7	8
1.6470652E-04	3.9705139E-04	5.5265577E-05	2.0242247E-05	1.6887048E-05	6.3639566E-06	6.4133909E-06	3.3613411E-06
9	10	11	12	13	14	15	16
4.5048434E-06	4.3384792E-06	1.1541980E-06	1.7279871E-06	1.3697599E-06	7.0342226E-07	4.9938405E-07	3.0470681E-07
17	18	19					
1.3849805E-06	8.2681476E-07	1.1737362E-06					

RADIAL SHELL DEFLECTION SPECTRA

1	2	3	4	5	6	7	8
1.140714E-08	2.9347816E-08	4.1184472E-08	4.7039669E-08	4.2115841E-08	3.0690050E-08	1.2288300E-08	3.0730771E-08

RADIAL SHELL ACCELERATION SPECTRA

1	2	3	4	5	6	7	8
1.8917141E-01	4.8059842E-01	6.7443494E-01	7.7031935E-01	6.8968699E-01	5.0257878E-01	2.0123261E-01	5.0324563E-01

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FREQUENCY = 5.520000E 02

WARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.0576287E-15	1.4709450E-15	1.5161350E-15	1.6975438E-15	1.6232337E-15	1.0753047E-15	1.7941055E-15	2.9684083E-15	1.0576287E-15	1.4709450E-15	1.5161350E-15	1.6975438E-15	1.6232337E-15	1.0753047E-15	1.7941055E-15	2.9684083E-15
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
2.4684510E-15	2.6441061E-15	3.5378852E-15	5.1725320E-15	7.4211518E-15	1.0806884E-14	1.5444730E-14	2.5556476E-14	2.4684510E-15	2.6441061E-15	3.5378852E-15	5.1725320E-15	7.4211518E-15	1.0806884E-14	1.5444730E-14	2.5556476E-14
17	17	18	18	19	19										
8.6360157E-14	2.7196107E-13	3.3801177E-13													

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.0253775E-06	1.4347198E-06	1.4699022E-06	1.6360841E-06	1.5737351E-06	1.0425145E-06	1.7393963E-06	2.8778900E-06	1.0253775E-06	1.4347198E-06	1.4699022E-06	1.6360841E-06	1.5737351E-06	1.0425145E-06	1.7393963E-06	2.8778900E-06
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
2.3931784E-06	2.5634770E-06	3.4300013E-06	5.0148014E-06	7.1948522E-06	1.0477340E-05	1.4973761E-05	2.4777161E-05	2.3931784E-06	2.5634770E-06	3.4300013E-06	5.0148014E-06	7.1948522E-06	1.0477340E-05	1.4973761E-05	2.4777161E-05
17	17	18	18	19	19										
8.2747501E-05	2.6366793E-04	3.2770448E-04													

RADIAL SHELL DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
4.0837260E-10	7.0973048E-10	5.6542163E-10	5.3670401E-10	5.7820357E-10	7.4218837E-10	4.3671654E-10	7.0378881E-10	4.0837260E-10	7.0973048E-10	5.6542163E-10	5.3670401E-10	5.7820357E-10	7.4218837E-10	4.3671654E-10	7.0378881E-10

RADIAL SHELL ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
3.9531974E-01	6.8808806E-01	5.4317975E-01	5.2033783E-01	5.6057191E-01	7.1955618E-01	4.2339937E-01	6.8232756E-01	3.9531974E-01	6.8808806E-01	5.4317975E-01	5.2033783E-01	5.6057191E-01	7.1955618E-01	4.2339937E-01	6.8232756E-01

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FREQUENCY = 5.550000E 02

DARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.031534JF-15	1.4412214E-15	1.4788125E-15	1.6412399E-15	1.5538239E-15	1.0427409F-15	1.7168013E-15	2.8658977E-15								
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
2.3232538E-15	2.5005327E-15	3.3443300E-15	4.9251912E-15	6.8831377E-15	1.0102745E-14	1.4304091E-14	2.2322511E-14								
17	17	18	18	19	19										
7.3532814E-14	2.1990196E-13	2.6882591E-13													

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.0219973E-06	1.4274970E-06	1.4551406E-06	1.6260663E-06	1.5394585E-06	1.0331005E-06	1.7009292E-06	2.8394019E-06								
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
2.3711277E-06	2.4863316F-06	3.3134110E-06	4.8796567E-06	6.8195018E-06	1.0009343E-05	1.4171847E-05	2.2116135E-05								
17	17	18	18	19	19										
7.2852988F-05	2.1786892F-04	2.6634056E-04													

RADIAL SHELL DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
3.3867053F-10	5.7909798E-10	4.6091598E-10	4.4934135E-10	4.7133567E-10	6.0558164E-10	3.6217659E-10	5.7325646E-10								

RADIAL SHELL ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
3.3553945F-01	5.7374411E-01	4.5665472E-01	4.4419634E-01	4.6697809E-01	5.9998292E-01	3.5882820E-01	5.6795658E-01								

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NARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.0049742E-15	1.4021091E-15	1.4408631E-15	1.5947531E-15	1.4869606E-15	1.0101645E-15	1.6422281E-15	1.6422281E-15	1.4869606E-15	1.0101645E-15	1.6422281E-15	1.6422281E-15	1.4869606E-15	1.0101645E-15	1.6422281E-15	1.6422281E-15
2.3179453E-15	2.3817544E-15	3.1614022E-15	4.6889271E-15	6.3898506E-15	9.4495405E-15	1.3262583E-14	1.3262583E-14	6.3898506E-15	9.4495405E-15	1.3262583E-14	1.3262583E-14	6.3898506E-15	9.4495405E-15	1.3262583E-14	1.3262583E-14
6.3776977E-14	1.8050199E-13	2.1746931E-13													

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.0173865E-06	1.4194264E-06	1.4586589E-06	1.6144497E-06	1.5053258E-06	1.0226409E-06	1.6625647E-06	1.6625647E-06	1.4586589E-06	1.0226409E-06	1.6625647E-06	1.6625647E-06	1.4586589E-06	1.0226409E-06	1.6625647E-06	1.6625647E-06
2.3455739E-06	2.4111711E-06	3.2004482E-06	4.7468393E-06	6.4687707E-06	9.5662504E-06	1.3426387E-05	1.3426387E-05	6.4687707E-06	9.5662504E-06	1.3426387E-05	1.3426387E-05	6.4687707E-06	9.5662504E-06	1.3426387E-05	1.3426387E-05
6.4564677E-05	1.8273134E-04	2.2015252E-04													

RADIAL SHELL DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
2.8505111E-10	4.8030150E-10	3.8184961E-10	3.7958305E-10	3.9048208E-10	5.0226689E-10	3.0483556E-10	3.0483556E-10	3.9048208E-10	5.0226689E-10	3.0483556E-10	3.0483556E-10	3.9048208E-10	5.0226689E-10	3.0483556E-10	3.0483556E-10
6.4564677E-05	1.8273134E-04	2.2015252E-04													

RADIAL SHELL ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
2.8857173E-01	4.8623363E-01	3.8556578E-01	3.8427122E-01	3.9530486E-01	5.0847031E-01	3.0860054E-01	3.0860054E-01	3.9530486E-01	5.0847031E-01	3.0860054E-01	3.0860054E-01	3.9530486E-01	5.0847031E-01	3.0860054E-01	3.0860054E-01
6.4564677E-05	1.8273134E-04	2.2015252E-04													

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DARC PRGM

NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
9.7304849E-16	1.3626444E-15	1.4022991E-15	1.5480766E-15	1.4229397E-15	9.7766680E-16	1.5708135E-15	1.5708135E-15	1.5708135E-15	1.5708135E-15	1.5708135E-15	1.5708135E-15	1.5708135E-15	1.5708135E-15	1.5708135E-15	1.5708135E-15
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
2.2425398E-15	2.2610729E-15	2.9889702E-15	4.4628512E-15	5.9387387E-15	8.8417321E-15	1.2309277E-14	1.2309277E-14	1.2309277E-14	1.2309277E-14	1.2309277E-14	1.2309277E-14	1.2309277E-14	1.2309277E-14	1.2309277E-14	1.2309277E-14
17	17	18	18	19	19										
5.5664752E-14	1.5012739E-13	1.7359970E-13													

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.0115936E-06	1.4093804E-06	1.4503952E-06	1.6011726E-06	1.4717438E-06	1.0111988E-06	1.6246894E-06	1.6246894E-06	1.6246894E-06	1.6246894E-06	1.6246894E-06	1.6246894E-06	1.6246894E-06	1.6246894E-06	1.6246894E-06	1.6246894E-06
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
2.3194545E-06	2.3386232E-06	3.0914860E-06	4.6159182E-06	6.1424257E-06	9.1449862E-06	1.2731461E-05	1.2731461E-05	1.2731461E-05	1.2731461E-05	1.2731461E-05	1.2731461E-05	1.2731461E-05	1.2731461E-05	1.2731461E-05	1.2731461E-05
17	17	18	18	19	19										
5.7573944E-05	1.5527766E-04	1.8472532E-04													

RADIAL SHELL DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
2.4302067E-10	4.0409709E-10	3.2084954E-10	3.2512855E-10	3.2810312E-10	4.2257743E-10	2.5988789E-10	2.5988789E-10	2.5988789E-10	2.5988789E-10	2.5988789E-10	2.5988789E-10	2.5988789E-10	2.5988789E-10	2.5988789E-10	2.5988789E-10

RADIAL SHELL ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
2.5135580E-01	4.1795683E-01	3.3185405E-01	3.3627982E-01	3.3935641E-01	4.3707101E-01	2.6880153E-01	2.6880153E-01	2.6880153E-01	2.6880153E-01	2.6880153E-01	2.6880153E-01	2.6880153E-01	2.6880153E-01	2.6880153E-01	2.6880153E-01

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NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
0.5095025E-16	1.3229524E-15	1.3631448E-15	1.5012138E-15	1.3619833E-15	9.4532920E-16	1.5025950E-15	2.5662426E-15	1.5025950E-15	2.5662426E-15	1.5025950E-15	2.5662426E-15	1.5025950E-15	2.5662426E-15	1.5025950E-15	2.5662426E-15
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
2.1670732E-15	2.1476589E-15	2.8268160E-15	4.2462050E-15	5.5272059E-15	8.2746346E-15	1.1434755E-14	1.5375965E-14	1.1434755E-14	1.5375965E-14	1.1434755E-14	1.5375965E-14	1.1434755E-14	1.5375965E-14	1.1434755E-14	1.5375965E-14
17	17	18	18	19	19	20	20	21	21	22	22	23	23	24	24
4.8876114E-14	1.2630808E-13	1.4862346E-13													

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.7046685E-06	1.3978317E-06	1.4402990E-06	1.5861827E-06	1.4390717E-06	9.9883497E-07	1.5876421E-06	2.7114923E-06	1.5876421E-06	2.7114923E-06	1.5876421E-06	2.7114923E-06	1.5876421E-06	2.7114923E-06	1.5876421E-06	2.7114923E-06
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
2.2997299E-06	2.2692165E-06	2.9868142E-06	4.4865407E-06	5.8400465E-06	8.7429801E-06	1.2081964E-05	1.6246247E-05	1.2081964E-05	1.6246247E-05	1.2081964E-05	1.6246247E-05	1.2081964E-05	1.6246247E-05	1.2081964E-05	1.6246247E-05
17	17	18	18	19	19	20	20	21	21	22	22	23	23	24	24
5.1642510E-05	1.3345714E-04	1.5703557E-04													

RADIAL SHFL DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
2.0949845E-10	3.4421703E-10	2.7291657E-10	2.8131376E-10	2.7908660E-10	3.5995889E-10	2.2403898E-10	3.3924004E-10	2.2403898E-10	3.3924004E-10	2.2403898E-10	3.3924004E-10	2.2403898E-10	3.3924004E-10	2.2403898E-10	3.3924004E-10

RADIAL SHFL ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
2.2135609E-01	3.6369976E-01	2.8836369E-01	2.9723615E-01	2.9488294E-01	3.8033261E-01	2.3671962E-01	3.5844108E-01	2.3671962E-01	3.5844108E-01	2.3671962E-01	3.5844108E-01	2.3671962E-01	3.5844108E-01	2.3671962E-01	3.5844108E-01

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NORMAL COORDINATE DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
0.4173890E-16	1.3096914E-15	1.3499690E-15	1.4855544E-15	1.3423722E-15	9.3459816E-16	1.4806013E-15	2.5338103E-15	15	16	16	16	15	15	16	16
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
2.1419148E-15	2.1114811E-15	2.7745989E-15	4.1759670E-15	5.3983684E-15	8.0939416E-15	1.1159309E-14	1.4791719E-14	17	17	18	18	19	19	19	19
4.6862783E-14	1.1951344E-13	1.4016114E-13													

NORMAL COORDINATE ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.0021173E-06	1.3936606E-06	1.4365206E-06	1.5807988E-06	1.4284367E-06	9.9451873E-07	1.5755281E-06	2.6962623E-06	12	12	14	14	15	15	16	16
9	9	10	10	11	11	12	12	13	13	14	14	15	15	16	16
2.2792410E-06	2.2468560E-06	2.9529144E-06	4.4437038E-06	5.7444779E-06	8.6128743E-06	1.1874774E-05	1.5740071E-05	17	17	18	18	19	19	19	19
4.9967331E-05	1.2717589E-04	1.4914739E-04													

RADIAL SHELL DEFLECTION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.0992478E-10	3.2709869E-10	2.5921497E-10	2.6860370E-10	2.6507527E-10	3.4205768E-10	2.1369388E-10	3.2223114E-10	17	17	18	18	19	19	19	19
1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8

RADIAL SHELL ACCELERATION SPECTRA

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
2.1263629E-01	3.4807021E-01	2.7583421E-01	2.8582488E-01	2.8207024E-01	3.6398828E-01	2.2739460E-01	3.4289058E-01	17	17	18	18	19	19	19	19
1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8

MEAN-SQUARE NORMAL COORDINATE DEFLECTION

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.147392E-08	5.3457134E-09	1.9972329E-09	8.7386532E-10	7.0383907E-10	5.9579210E-10	2.2000951E-10	2.1105466E-10	9	9	10	10	11	11	12	12
1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
1.3341757E-10	1.5348502E-10	3.6813468E-11	7.5285412E-11	6.7398306E-11	3.2059650E-11	5.0348374E-11	2.1317996E-11	17	17	18	18	19	19	19	19
1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8

MEAN-SQUARE NORMAL COORDINATE ACCELERATION

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
9.1264791E-02	9.4929081E-02	5.9106933E-02	3.2354175E-02	2.5670937E-02	2.2794099E-02	7.9737490E-03	4.4869198E-03	9	9	10	10	11	11	12	12
1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
4.2107676E-03	3.3523036E-03	2.1712457E-03	3.3303818E-03	5.1402037E-03	7.2223434E-03	9.0577172E-03	1.0103133E-02	17	17	18	18	19	19	19	19
1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
3.7456935E-02	3.5698780E-02	3.6761987E-02													

MEAN-SQUARE RADIAL DEFLECTION

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
6.5654225E-06	1.5886402E-05	2.1378777E-05	2.4327089E-05	2.1862242E-05	1.6612969E-05	7.0211314E-06	1.6539146E-05	17	17	18	18	19	19	19	19
1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8

MEAN-SQUARE RADIAL ACCELERATION

1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8
2.2905408E-02	3.9935088E-02	3.7337343E-02	4.3263065E-02	3.8181627E-02	4.0715699E-02	2.4495192E-02	3.8754669E-02	17	17	18	18	19	19	19	19
1	1	2	2	3	3	4	4	5	5	6	6	7	7	8	8

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