

ESTIMATING REFRACTIVE INDEX SPECTRA IN REGIONS
OF CLEAR AIR TURBULENCE

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PREFACE

This is the first technical report prepared under NASA Grant NGR 44-012-048. It is intended to serve as a background in the meteorological aspects of the problem for further theoretical and experimental studies on the technique for detection of CAT.

The second report which will follow in a few weeks will be entitled "Remote Detection of CAT" by Dr. B. M. Fannin. This second report will examine the existing theories on radar return from disturbed refractive index regions.

A third report which is in the process of preparation will be concerned with equipment for direct measurement of refractive index differences and the examination of the initial data taken on a 270 foot tower. The results of these measurements should shed considerable light on the nature of refractive index anomalies associated with refractive index variation of the atmosphere.

A fourth aspect of the research is concerned with the use of radar for measuring the return from the refractive index variation in the atmosphere. Analysis of experimental programs is currently under way. Preliminary proposals have been made to the Electronics Research Center for unique tests which would measure scattering from radar beams and the associated refractive index anomalies. It is hoped that this program can be continued in an extension of the grant period.

A. W. Straiton
Principal Investigator

CONCLUSIONS

~~As suggested by the cumulative distribution shown in Fig. 1 and the summary shown in Table 2,~~ clear-air turbulence patches average from 500 to 3000 feet in depth and one-half have horizontal dimensions less than 20 miles.

Clear-air turbulence occurs preferentially on the boundaries of shallow, thermally-stable baroclinic zones in the vicinity of the jet stream. The average percentage frequency in the jet stream vicinity is shown in Fig. 2. A schematic presentation of the variation with flow curvature is given in Fig. 4. These indicate a preference for occurrence on the cyclone side of the jet core and topographic effects may enhance these frequencies.

The refractive index spectrum for scales in the inertial subrange in the presence of clear-air turbulence can be expressed in terms of the mechanical energy. For stable stratification, the dependence suggested by Tatarski [61] may be used to provide consistent estimates.

The refractive index spectral estimates for dry air fluctuations made here are less than those found by other investigators (for example, Tatarski [61], Atlas et al. [2]) by more than 10 db. Since the scattering at any angle is directly proportional to the spectral amplitude, the radar return will be correspondingly less than previous estimates.

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ESTIMATING REFRACTIVE INDEX SPECTRA IN REGIONS
OF CLEAR-AIR TURBULENCE

J. J. Stephens

E. R. Reiter¹

I. INTRODUCTION

Clear-air turbulence (CAT) may be defined as bumpiness in flight through clear air, i. e., away from clouds and at heights exceeding 15, 000 feet MSL. The turbulent mechanical energy associated with CAT must be of sufficient magnitude at the proper wavelengths to affect an aircraft flying in the region. Low-level turbulence induced mechanically by surface roughness is excluded from consideration, as is turbulence encountered in deep convective currents associated with cumulus clouds.

The problem under study here is the assessment of the spectrum of refractive index fluctuations to be expected in regions of moderate to severe CAT. Scattering of electromagnetic waves by turbulent inhomogeneities in refractive index will be treated in a later report by B. M. Fannin. Taken together, these constitute an alternate solution to the approach by Atlas, Hardy and Naito [2].

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II. LOCATIONS AND DIMENSIONS OF CAT

Clear-air turbulence is found in the upper troposphere and stratosphere, principally in shallow layers in the jet stream vicinity, in waves leeward of mountain ranges, and to a lesser extent in association with upper troughs and lows. Although light turbulence may be expected in layers of nearly adiabatic stratification, moderate to severe CAT appears to occur preferentially on the boundaries of thermally stable, baroclinic zones.

Various descriptive terms have been used to characterize the intensity of CAT. Criteria developed by the NACA Subcommittee on Meteorological Problems are shown in Table 1 along with objective criteria for use with the VGH meter² (Colson [15], Endlich [20]). The derived vertical, sharp-edge equivalent gust depends upon the aircraft response characteristics, as well as the atmospheric turbulent energy spectrum (see, for example, Reiter [46]). This led to a modified VGH instrument measuring horizontal gusts (Endlich [20]).

The horizontal extent of clear-air turbulence patches encountered in Project Jet Stream flights is shown in Fig. 1, after Cunningham [17]. More than 50% of the turbulent patches were less than 20 miles in extent. Nearly one-third were less than 10 miles. Cunningham considered that

² This instrument records velocity (air speed), gusts (acceleration), and height.

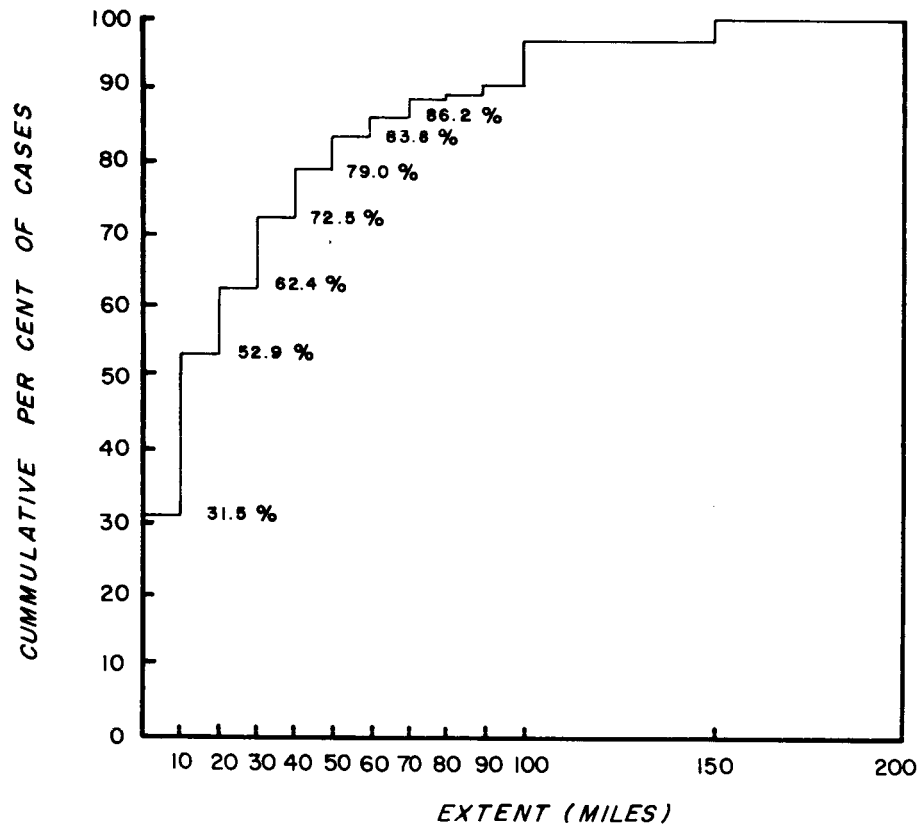


FIG. 1. HORIZONTAL EXTENT OF CLEAR-AIR
TURBULENCE ENCOUNTERED IN PROJECT
JET STREAM (*Cunningham [17]*)

Adjective	Description	Derived Gust Velocity		Airspeed
		Vertical feet per second	Horizontal feet per second	Fluctuations knots
Light	A turbulent condition during which occupants may be required to use seat belts, but objects in the aircraft remain at rest.	5-20	5-9	5-15
Moderate	A turbulent condition in which occupants require seat belts and occasionally are thrown against the belt. Unsecured objects in the aircraft move about.	20-35	10-19	15-25
Severe	A turbulent condition in which the aircraft momentarily may be out of control. Occupants are thrown violently against the belt and back into the seat. Objects not secured in the aircraft are tossed about.	35-50	>20	more than 24
Extreme	A rarely encountered turbulent condition in which the aircraft is violently tossed about, and is practically impossible to control. May cause structural damage.	over 50		rapid fluctuations over 25

Table 1. Turbulence criteria

other investigations had suggested a larger percentage for this smallest scale. Estoque [23] found a distribution of small patch sizes similar to Cunningham's results. Patches of larger dimensions are found less frequently and seem to be elongated in the direction of flow.

As suggested by the summary shown in Table 2 (Reiter [47]), the average vertical thicknesses of turbulent layers appear to lie between

1 Source	2 Altitude Range	3 Alt. Range of Frequency Max.	4 Intensity of Turbulence	5 Frequency of Occurrence	6 Extreme Intensity	7		8 Extent of Turbulent Region		10
						Mean	Extreme	Mean	Extreme	
1. Bannon [3, 4]		28000 to 32000 ft.			0.7 g					
2. Bindon [5]	24000 ft.				3 g					
3. Clem [10]	25000 to 45000 ft.	40000 to 44000 ft. Secondary max. at 34000 over NE USA	light CAT moderate severe	19% of all flights 12% 2%		15 to 60 km, longer than wide		500 to 2000 ft.		
4. Clodman [11]	18000 to 40000 ft.	No significant altitude dependency over S. Canada betw. 18000 and 38000 ft.	all intensities moderate and severe	l: 35 km l: 85 km l: 440 km		~ 90 km	≥ 450 km	3000 ft.	100 to 1500 ft.	
5. Cunningham [17]						50% < 30 km	6. 9 to 328 km	~ 1500 ft.	500 to 3000 ft.	
5. Estoque [23]	150 mb 200 mb 250 mb 300 mb 350 mb		10 5 2 ft/sec 10 5 2 10 5 2 10 5 2 10 5 2	0.02 0.16 2.0% 0.09 0.75 9.0 0.15 1.2 15 0.11 0.88 11 0.08 0.72 9.0 of km flown						
7. Heath-Smith [24]		Decreasing to 25000 ft. above increasing again			0.8 g or 20 ft/sec					
9. Hislop [25]	25000 ft. 35000 ft.	250 mb (Europe, Turkey, USA, Japan)	4 ft/sec 8 ft/sec 12 ft/sec	1 gust per 13 km 1 gust per 97 km 1 gust per 650 km		75 to 100 km		900 m		
9. Hyde [26]	to 49000 ft. to 36000 ft. 22000 to 28000 ft. 40000 ft.		light or mod. severe extremely severe	28% of all flights 7% of all flights 1% of all flights	2.5 to 4 g (estimated)					
10. Kuettner [30]					+3 g, -2.5 g (sail plane) (21 m./sec)					
11. Murray [38]	400 to 200 mb		≥ 0.1 g ≥ 0.2 g ≥ 0.3 g	37.4% of all flights 4.9% of all flights 0.2% of all flights	0.4 g					
12. Pinus [42]		Decrease to middle troposph., above increase by 15 to 20% to tropopause								
13. Press et al. [43]						50% < 80 km Areas with strong CAT	> 800 km	50% < 2000 ft.	100 to 10000 ft.	

Table 2. Characteristics of Various CAT Observations

500 to 3000 feet. Clodman [11] reports an extreme case in which the turbulent layer extended over 15,000 feet. It is questionable, however, whether the turbulence was uniformly distributed, or consisted of a series of individual shallow turbulent layers with relatively quiescent intervening layers.

Frequency of occurrence statistics derived from turbulence encounters by aircraft depend upon season, location, type of aircraft, flight planning procedures, and the degree of subjectivity of the reports. Some of the results from early investigations are summarized in Table 2. Clodman, Morgan and Ball [12] have concluded that 3% of flight time over continental areas between 20,000 to 45,000 feet will be in turbulence. Of the 3% total, light turbulence may be expected 75% of the time, moderate 15-20%, severe 5-10% and extreme 1-3%. A somewhat larger percentage was found by Briggs and Roach [6]. Slight intensity was found 5.4% of flying time, moderate 4.0%, and severe 0.2%. Endlich and McLean [18] found a comparable result. Endlich [20] found CAT about 12% of the time. Of this, 2.5% was moderate and 0.2% was severe. Endlich and McLean [21] have determined the average distribution of light or greater intensity with respect to the jet stream core. As shown in Fig. 2, the greatest frequency is on the cyclonic side of the core, as noted by a number of investigators.

An idealized jet stream model cross section is shown in Fig. 3, after Endlich and McLean [18]. More complex models have been presented, for example, by Brundidge and Goldman [8] and Serebreny, Wiegman and

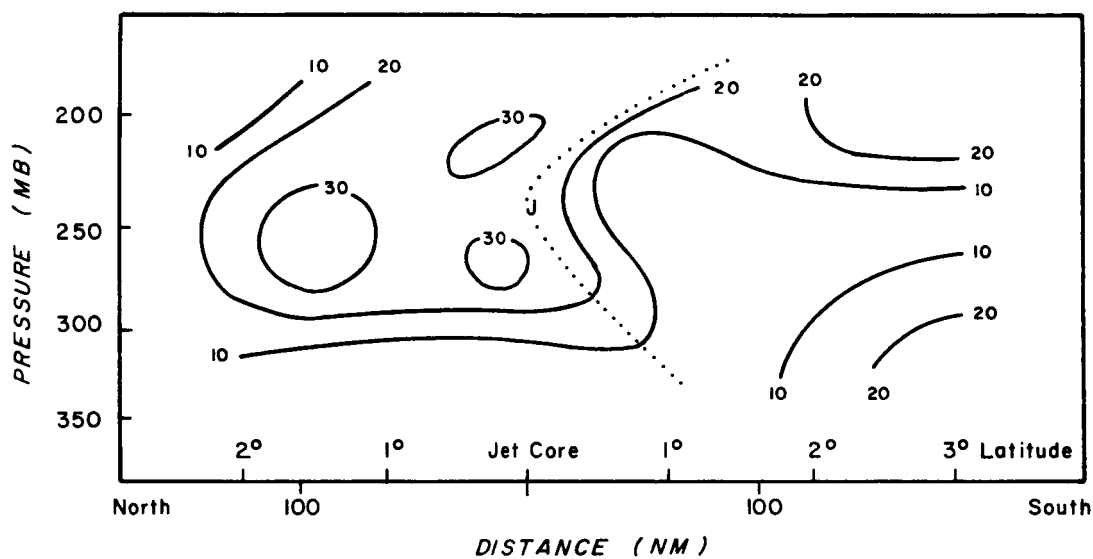


FIG. 2. AVERAGE DISTRIBUTION OF PER CENT OF TIME TURBULENCE OF LIGHT OR GREATER INTENSITY IS ENCOUNTERED IN THE JET STREAM. (after Endlich and McLean [21]).

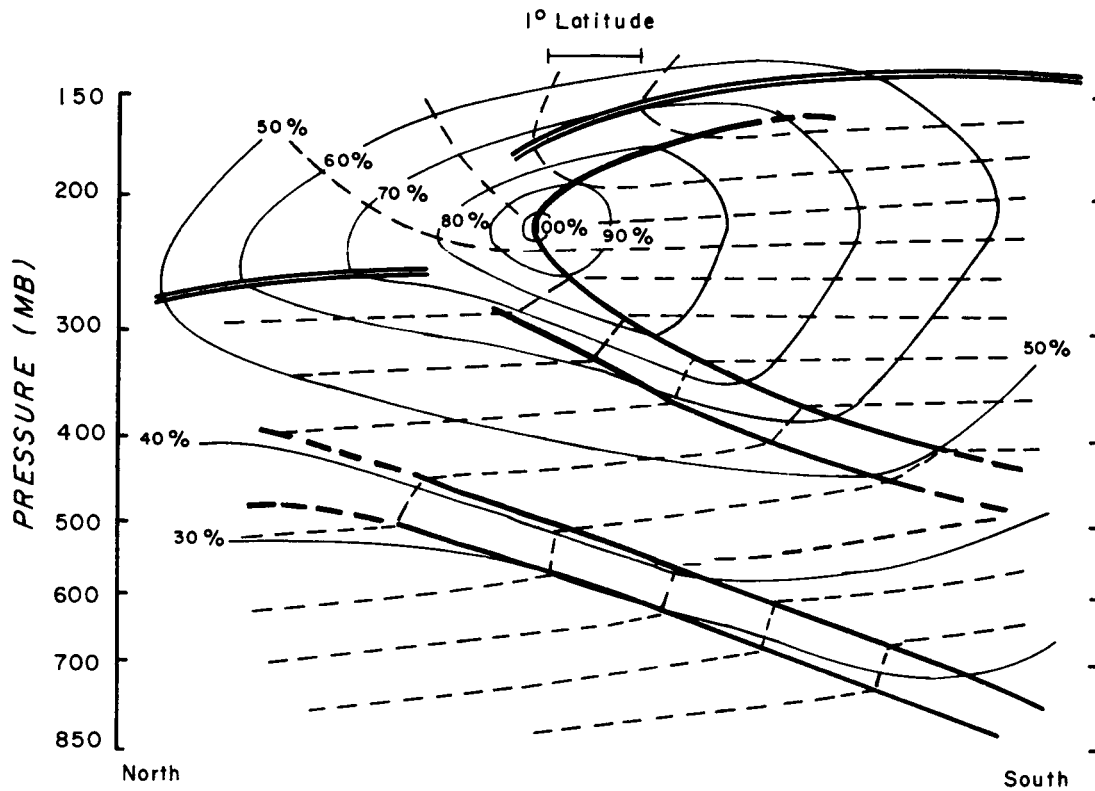


FIG. 3. IDEALIZED JET STREAM MODEL CROSS SECTION. ISOTACHS (% OF CORE SPEED) DENOTED BY THE SOLID LINE (—), DASHED LINES (---), ARE SCHEMATIC ISOTHERMS. STABLE LAYERS ARE INDICATED BY THICK SOLID LINES (—), AND DOUBLE LINES, (==) DENOTE THE POLAR AND SUBTROPICAL TROPOPAUSES. (after Endlich and McLean [10])

Hadfield [57], the latter indicative of those proposed by Canadian workers. While the Endlich and McLean model is somewhat simplified and indicative only of the polar front jet, it does contain the stable, baroclinic zone (jet stream front). The polar front has been placed arbitrarily such that it intersects the 500-mb surface below the jet core. The core placement with respect to the jet stream front and the tropopause break may be too far north.

The distributions of CAT with respect to schematic cross sections taken through trough, ridge and straight flow conditions are shown in Fig. 4 after Endlich [19]. These show a tendency for moderate to severe CAT occurrence on the upper boundary of the jet stream front and on the sub-tropical tropopause.

Any anticipation of CAT encounters should consider topographical effects. A schematic mountain wave flow regime is shown in Fig. 5, after Jenkins [27]. Turbulence is to be expected in the rotor circulation and in layers (not shown) where the wave motion leads to a maximum wind shear. The indicated turbulent layer at the tropopause is misleading as to the extent of turbulence.

Mountain wave behavior is largely governed by the Lyra [34] parameter:³

$$L^2 = \frac{g}{\theta u^2} \frac{\partial \theta}{\partial z} \quad (2.1)$$

³The Scorer [56] parameter differs by a wind profile curvature term that is difficult to evaluate.

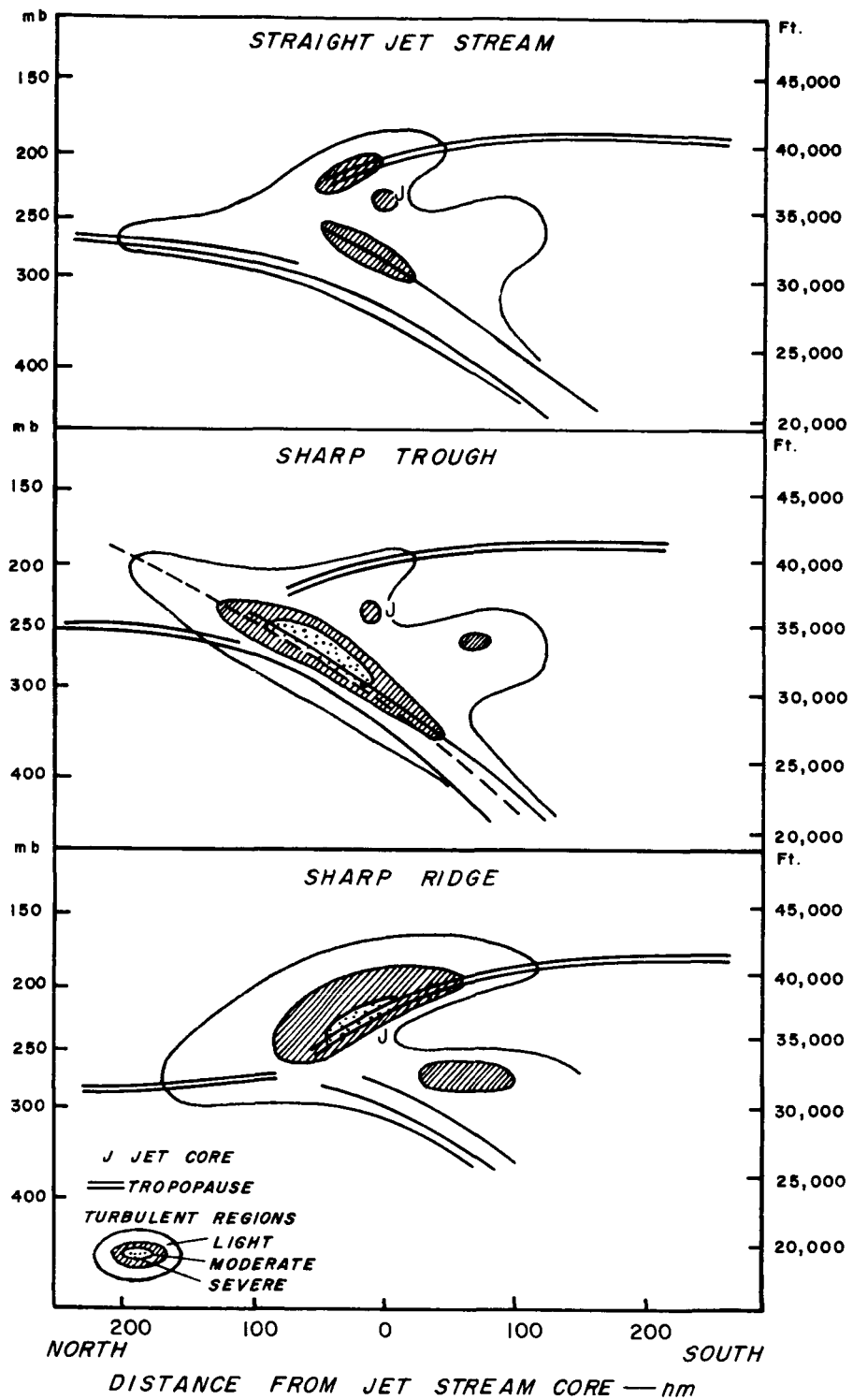


FIG. 4. IDEALIZED MODELS OF TURBULENT REGIONS IN THE VICINITY OF THE JET STREAM (after Endlich [19])

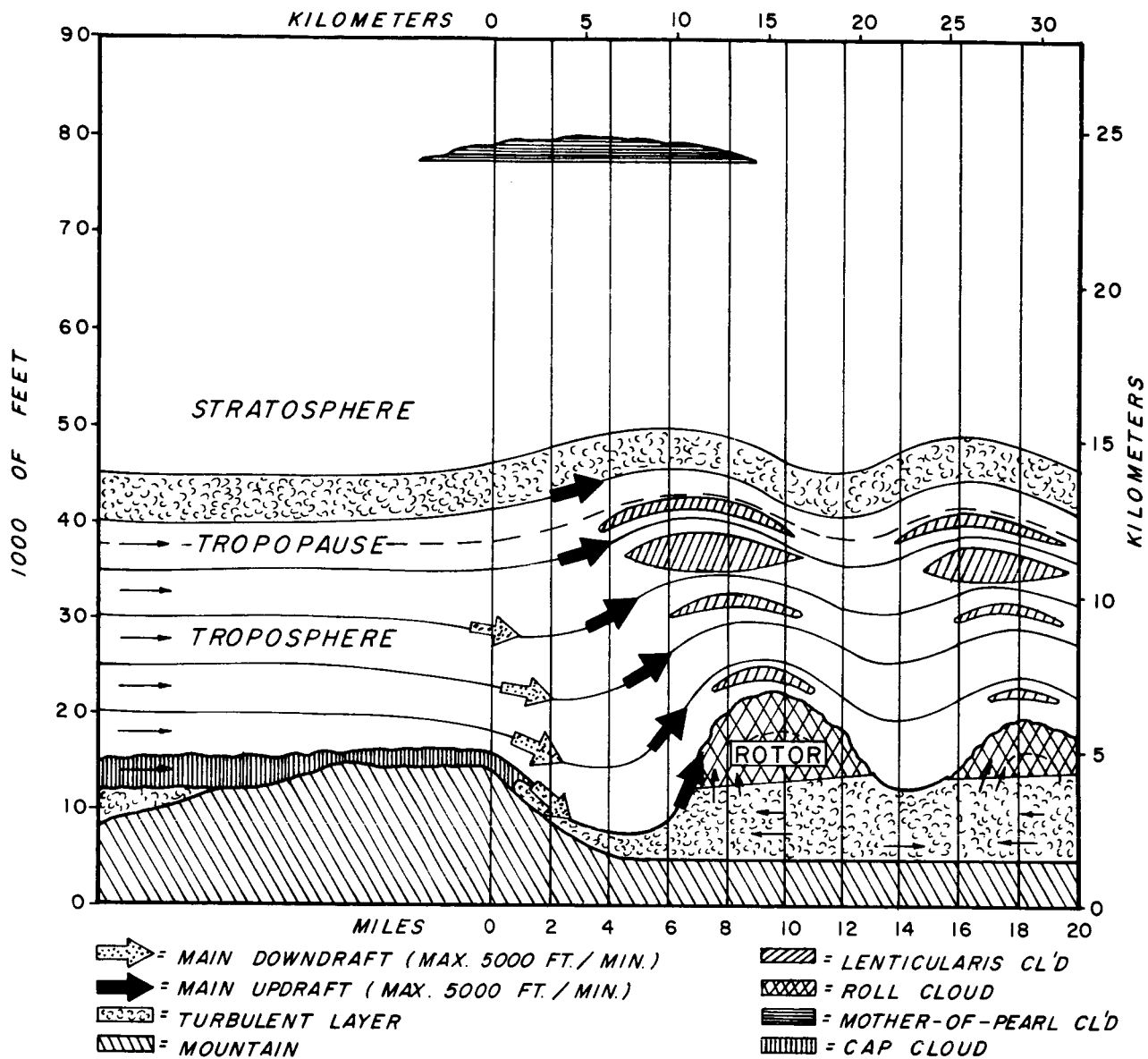


FIG. 5. CROSS SECTION OF CONDITIONS ASSOCIATED WITH A TYPICAL MOUNTAIN WAVE.

(after Jenkins [27])

where ℓ is the wave number generated in the lee of the mountain, g is the acceleration of gravity, θ is potential temperature, and u is wind speed component normal to the mountain range. The vertical coordinate is denoted by z . In a case study, Reiter [48] has shown that an aircraft loss in otherwise minimal turbulence conditions was likely due to a mountain wave phenomenon. The Lyra parameter decreased to the lower boundary of the stable baroclinic zone where it increased to a secondary maximum, a favorable condition for the formation of lee waves.

A cloud photogrammetric study by Reiter and Nania [49] has shown that both standing and traveling gravity waves ranging from 100 meters to 50 kilometers are found at jet stream levels in the lee of the Rocky Mountains. The shorter waves are in the range of clear-air turbulence. Similar scales were found by Conover [16].

Even moderate terrain relief can have a pronounced effect on CAT frequencies. In a study of CAT occurrences in the lee of the small ridge shown in Fig. 6, Clodman, Morgan and Ball [12] found an enhanced frequency of occurrence (Fig. 7) when the 850-mb flow was approximately normal to the ridge and a reduction (Fig. 8) when the flow was parallel. Further, the enhancement was favored by large low-level wind speeds and small wind shear in the middle troposphere. The enhancement under mountain-wave conditions is also reflected in the statistics compiled by Colson [13, 14].

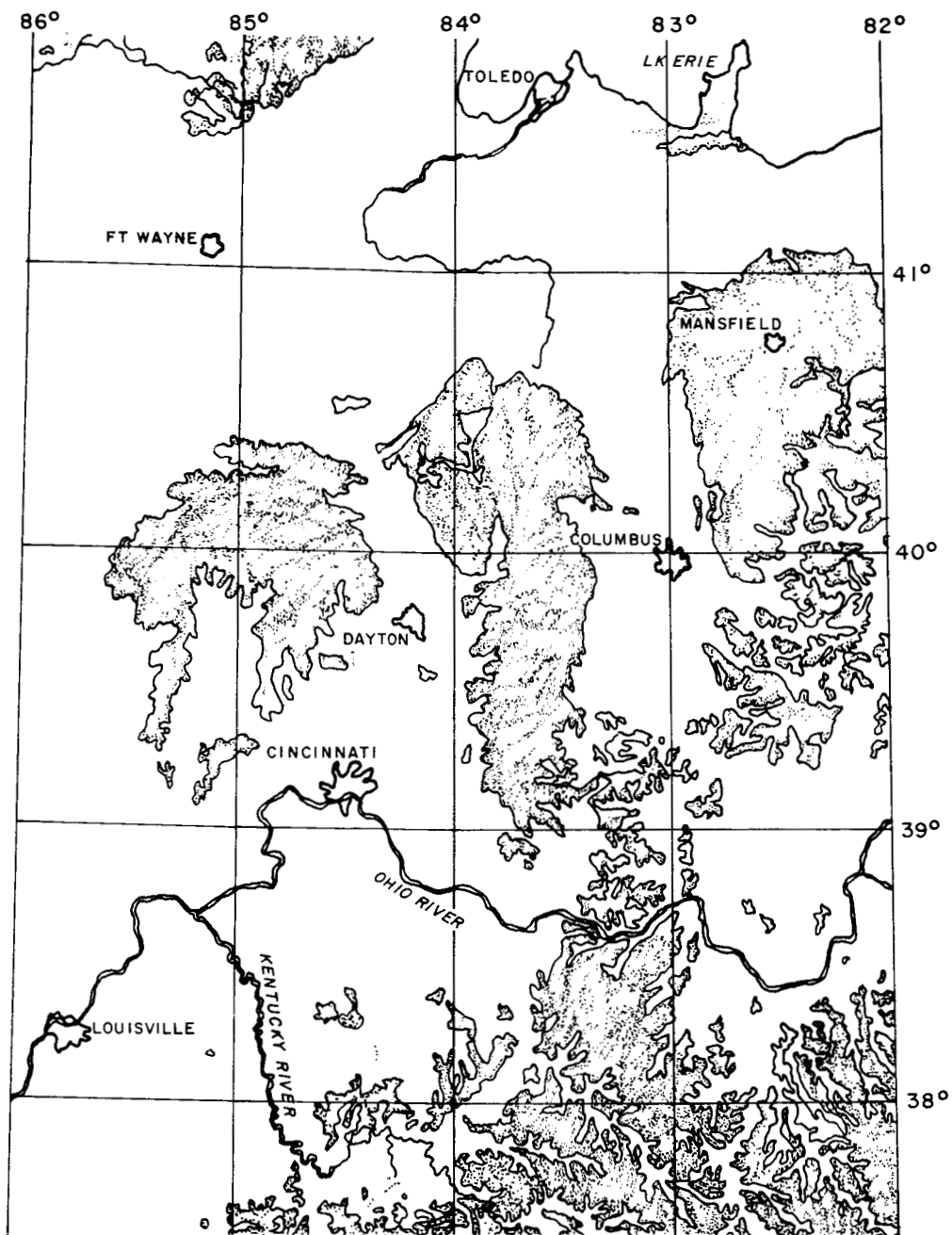


FIG. 6. TOPOGRAPHY IN VICINITY OF PATTERSON AFB.
UNSHADED AREAS 0-1000 FT. MSL ; SHADED AREAS
1000 - 2000 FT. MSL.

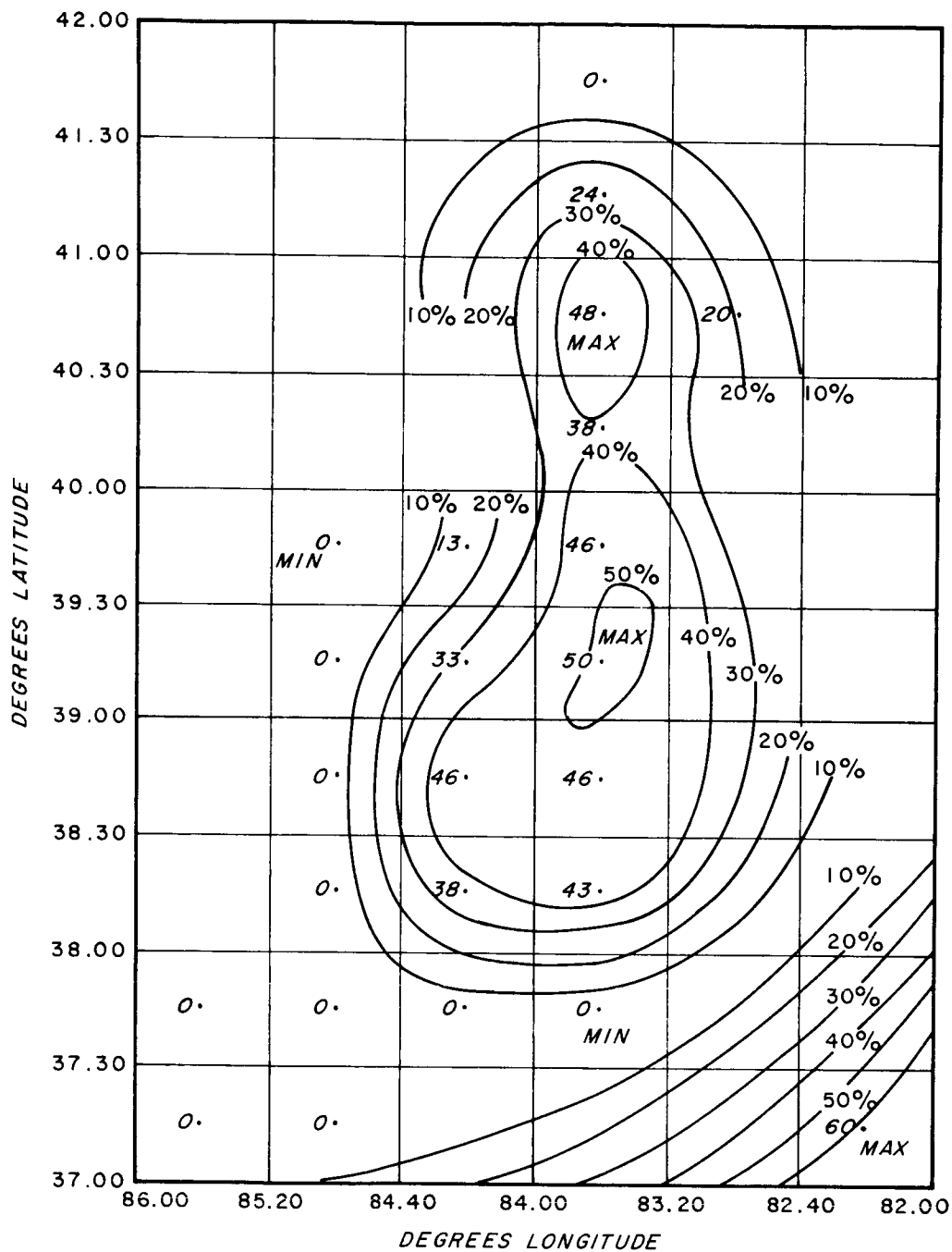


FIG. 7. PERCENTAGE OF TURBULENCE OCCURRENCES IN THE PATTERSON AREA WITH THE 850 MB CENTRAL WIND DIRECTION ($085-265^{\circ} \pm 30^{\circ}$) NORMAL TO THE CENTRAL RIDGE.

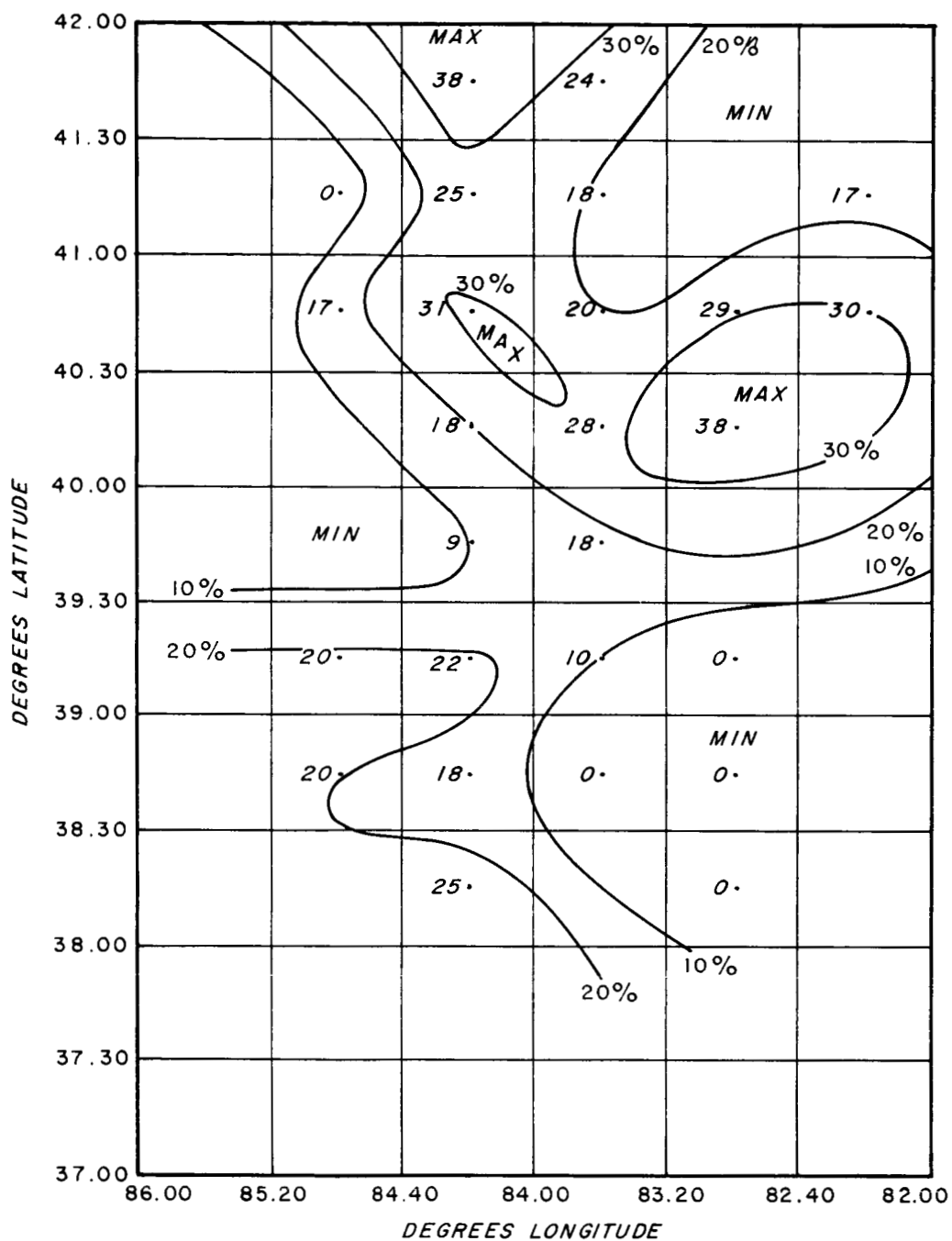


FIG.8. PERCENTAGE OF TURBULENCE OCCURRENCES IN THE PATTERSON AREA WITH THE 850 MB CENTRAL WIND DIRECTION ($175-355^\circ \pm 59^\circ$) PARALLEL TO THE CENTRAL RIDGE.

Finally, Clodman, Morgan and Ball [12] have shown that turbulence encounters over oceanic areas are an order of magnitude smaller (0.24%) than over continental areas.

III. MECHANISM, SIGNATURE AND STRUCTURE OF CAT

Turbulent fluctuations stemming from Reynolds stresses associated with a mean wind shear must work against the stabilizing effect of gravity. Richardson's well-known stability criterion asserts that turbulence can proceed only if turbulent generation exceeds the work against gravity. Thus, the flux Richardson number,

$$R_f = \frac{K_h}{K_m} \frac{\frac{g}{\theta} \frac{\partial \theta}{\partial z}}{\left(\frac{\partial \bar{u}_i}{\partial z} \right)^2}, \quad (3.1)$$

must be less than unity for the continuation of turbulence. Here \vec{u} is the mean fluid velocity and K_h and K_m are the eddy diffusivities for heat and momentum, respectively. A summation convention is implied for the shear tensor, $\partial \bar{u}_i / \partial x_j$.

As discussed by Lumley and Panofsky [33], the diffusivity ratio (K_h / K_m) is near unity for unstable stratification, but may decrease with increasing stability. The gradient Richardson number (= 1/Froude number) is obtained by multiplying R_f by the turbulent Prandtl number:

$$R_i = \left(\frac{K_h}{K_m} \right) R_f. \quad (3.2)$$

It is the gradient Richardson number that has been used with mixed success in correlation studies with CAT. Brundidge [7], Lake [31] and Reiter [45], as examples, found poor correlation between R_i and CAT. Anderson [1], Rustenbeck [53] and Kronebach [29] found some correlation. Petterssen and Swinbank [41] found that the Richardson criterion was applicable in the free atmosphere provided that $K_h/K_m \sim 0.65$. The disparity in results may be attributable to the vertical grid coarseness in the R_i -calculations.

Other criteria have been developed. Radok and Clarke [44] have given a modified (thermal) Richardson number. Sasaki [54] has derived a criterion involving wind and temperature curvatures for a three-layer model. The latter appears to be superior to the Richardson criterion for CAT, but suffers from the necessity of determining second vertical differences.

Other parameters have been studied with mixed or little success. Considerations of the stability, $\frac{g}{\theta} \partial\theta/\partial z$, alone have met with little success. The vertical wind shear has been suggested by some investigators as superior to the Richardson number. Also, a turbulence index (Endlich and Mancuso [22]), $|u \frac{\partial \alpha}{\partial z} \frac{\partial^2 T}{\partial z^2}|$, where α is wind direction and T is absolute temperature, was found to be as skillful as R_i in identifying turbulent regions. The horizontal shear has also been examined, again with mixed results. A study of two confluent jet streams by Reiter and Nania [49] has shown that the vector wind shear should be considered.

This is in accord with the results of Endlich and Mancuso [22], and the suggestions by Keitz [28] and Schwerdtfeger and Radok [55], except that the pertinent differential advection in the first study appears to be in the sense of atmospheric stabilization.

The criteria above are not scale-dependent. Since moderate and severe CAT appears to occur preferentially on the upper boundary of thermally stable layers, an appropriate consideration is the critical wavelength, λ_c , of gravitational shearing (Helmholtz) waves on the interface between two stably stratified, incompressible fluids (Reiter [47, 50]):

$$\lambda_c = \frac{2\pi}{g} \frac{(u_0 - u_1)^2 T_0 T_1}{(T_1 + T_0)(T_1 - T_0)}, \quad (3.3)$$

where the subscripts 0 and 1 refer to the lower and upper layers, respectively. Wavelengths less than λ_c are unstable. This is indicated schematically in Fig. 9. The large black arrows denote the wind vectors in the two layers. The gravity wave crests are normal to the vector shear. Small black arrows indicate vertical motion. Such gravity waves may be excited, for example, as lee waves. The wind shear necessary to produce certain critical wave lengths is shown in Table 3 for various temperature discontinuities (Reiter [50]). The value of λ_c will be reduced for finite transition layers, but the range of wavelengths can easily fall within the CAT spectrum input region.

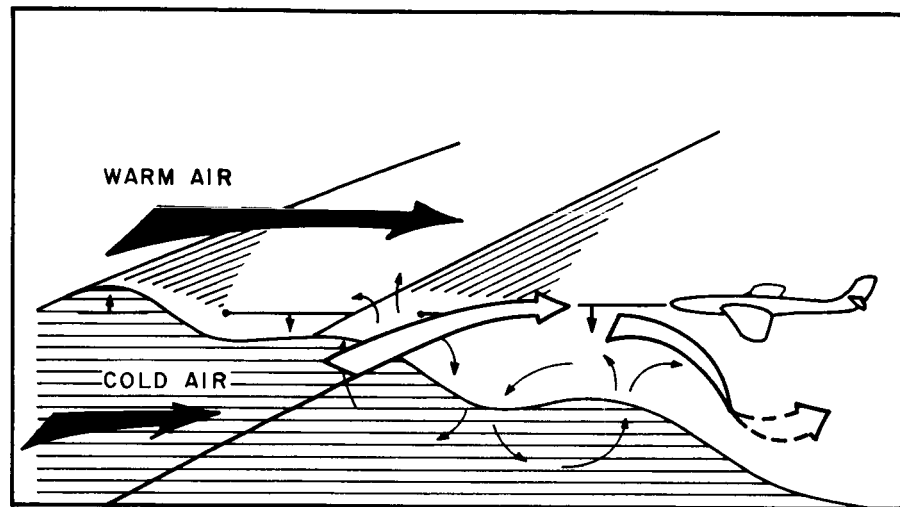


FIG. 9. SCHEMATIC DIAGRAM OF AIR FLOW IN THE VICINITY OF AN IDEALIZED GRAVITATIONAL SHEARING WAVE (*adapted from Reiter [51]*)

$\Delta T(^{\circ}\text{C})$	$\lambda_c = 200 \text{ m}$	$\lambda_c = 100 \text{ m}$	$\lambda_c = 50 \text{ m}$
2	2.3 m sec^{-1}	1.6	1.2
4	3.3	2.3	1.6
6	4.0	2.9	2.0
8	4.7	3.3	2.3
10	5.2	3.7	2.6

Table 3. Vertical wind shear for different temperature discontinuities and critical wave lengths at an interface.

The argument for the "breaking" of gravitational shearing waves as the energy source for CAT may be substantiated by an examination of mechanical energy spectra such as shown by Reiter and Burns [52]. Smoothed spectra of the u-, v- and w-components (along flight direction, transverse and vertical, respectively) are shown in Fig. 10 (after Burns and Rider [9]) for moderate to severe CAT. Since the same turbulence patch was studied for both the headwind and transverse measurements, Reiter and Burns [52] have argued that a wave phenomenon of specific orientation was responsible for the energy input at wavelengths corresponding to the "hump" in the spectra.

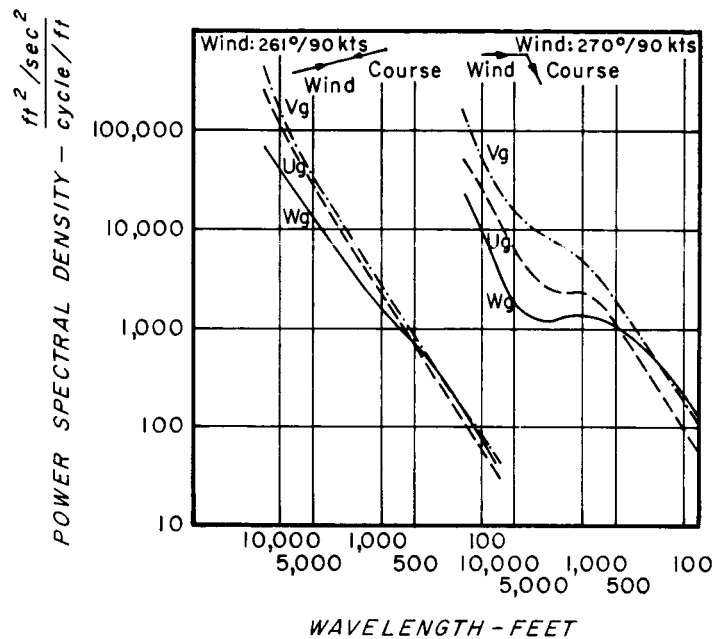


FIG. 10. "SMOOTHED" SPECTRA OF u -, v -, AND w -COMPONENTS OF TURBULENCE MEASURED DURING FLIGHTS 18B AND 18D (*after Burns and Rider, 1965 [9]*).

IV. REFRACTIVE INDEX SPECTRA

Estimates of the scattering from atmospheric turbulent fluctuations require an evaluation of the spectral density of refractive index at a wave number $4\pi \sin(\frac{\theta}{2})/\lambda$, where θ is the scattering angle and λ is the wave length of the radiation. For backscattering ($\theta=\pi$) this is twice the radian wave number associated with the incident radiation. As suggested schematically in Fig. 11, the scattering cross section is determined by a narrow band filter acting on the refractive index spectrum. The filtered evaluation has been placed near the dissipation subrange boundary. The implications are discussed below.

Following Tatarski [61], the fluctuation spectrum for the refractive index will be approached by considering conservative passive additives, say $X(\vec{r})$, where \vec{r} is a position vector. Although X will ultimately be taken as the potential refractivity, fluctuations in refractive index near the tropopause are due primarily to temperature variations and considerations for temperature spectra will be used as guidelines. In the considerations that follow, the Reynolds⁴ and Péclet^{5,6} numbers are taken to be large so that a convective subrange exists where the pertinent outer scale,

⁴ The dimensionless ratio of measures of inertia to viscous forces, R_e .

⁵ The dimensionless ratio of convective to molecular flux of heat, P . The Péclet and Reynolds numbers are related by $P = \sigma R_e$, where $\sigma (= \nu/\kappa)$ is the Prandtl number, the ratio of molecular momentum diffusivity to thermal diffusivity.

⁶ A more appropriate number would be sR_e , where $s (= \nu/\nu_x)$ is the Schmidt number, the ratio of molecular momentum diffusivity to x -diffusivity.

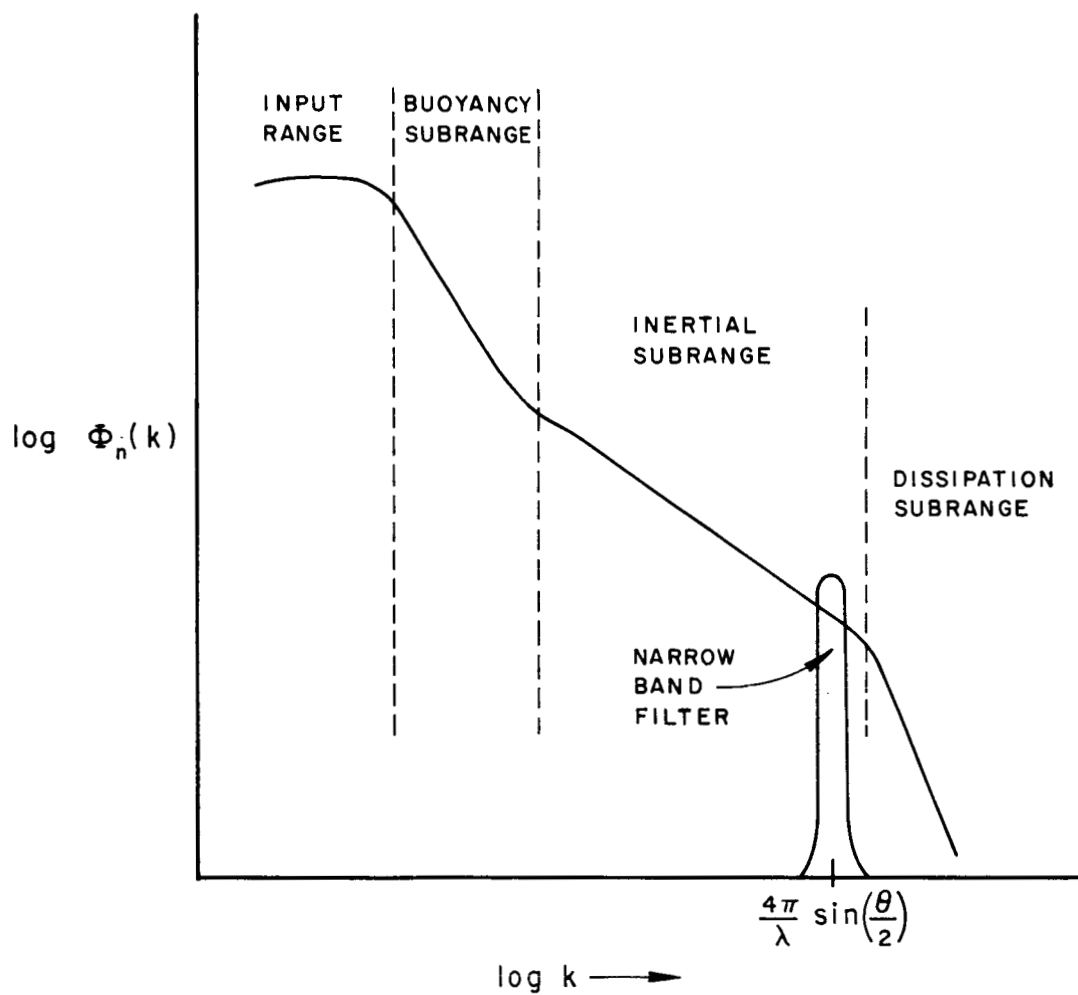


FIG. II. SCHEMATIC REFRACTIVE INDEX SPECTRUM AS A FUNCTION OF WAVE NUMBER INDICATING TYPICAL RANGES AND THE NARROW BAND FILTER WHICH DETERMINES SCATTERING FROM TURBULENT FLUCTUATIONS (adapted from Staras and Wheelon [58])

L_x , and the diffusive inner scale, ℓ_x , are such that $L_x \gggg \ell_x$.

In a locally isotropic random field the structure function

$$D_x(|\vec{r}_1 - \vec{r}_2|) = \overline{[X(\vec{r}_1) - X(\vec{r}_2)]^2} \text{ ensemble} \quad (4.1)$$

then depends only upon $r (= |\vec{r}_1 - \vec{r}_2|)$, η and ϵ where η represents the amount of inhomogeneity which disappears per unit time due to molecular diffusion and ϵ is the energy dissipation rate. In this convective, inertial subrange dimensional considerations show that

$$D_x(r) = \begin{cases} C_x^2 r^{2/3} & , \quad \ell_x \ll r \ll L_x & \text{(a)} \\ C_x^2 \ell_x^{2/3} \left(\frac{r}{\ell_x}\right)^2 & , \quad r \ll \ell_x & \text{(b)} \end{cases} \quad (4.2)$$

where

$$C_x^2 = a^2 \eta \epsilon^{-1/3} \quad (4.3)$$

Here the parameter a^2 depends upon the Richardson number for stable stratification. The inner and outer scales are given by Tatarski as

$$\ell_x = (27a^6 \nu_x^3 / \epsilon)^{1/4} \quad (4.4)$$

and

$$L_x = (K_x / |(\partial \bar{u}_i / \partial x_j)|)^{1/2} \quad (4.5)$$

where ν_x is the molecular diffusion coefficient for X and K_x is the coefficient of turbulent diffusion of X (Tatarski takes this as the

momentum coefficient). Here the inner scale is found as the point of intersection of the asymptotic expansions of eqs. 4.2(a) and 4.2(b).

The rate of molecular dissipation of x-fluctuations is given by

$$\eta = \overline{v_x (\text{grad } X')^2} = K_x (\text{grad } \bar{X})^2. \quad (4.6)$$

The energy dissipation rate is

$$\epsilon = K_m (\partial \bar{u}_i / \partial x_j)^2 \quad (4.7)$$

The three-dimensional spectral density corresponding to (4.2) is

$$\Phi_x(k) = \frac{\Gamma(\frac{8}{3}) \sin(\frac{\pi}{3})}{4\pi^2} C_x^2 k^{-11/3} \quad (4.8)$$

where k is radian wave number ($2\pi/\lambda$). Here it will be convenient to use the one-dimensional counterpart which may be defined by

$$\Phi_x(k) = -\frac{1}{2\pi k} \frac{dF_x(k)}{dk}, \quad (4.9)$$

or

$$F_x(k_1) = -2\pi \int_{-\infty}^{k_1} k \Phi_x(k) dk. \quad (4.10)$$

Thus,

$$F_x(k) = \frac{3}{10\pi} \Gamma(\frac{8}{3}) \sin(\frac{\pi}{3}) C_x^2 k^{-5/3}. \quad (4.11)$$

Equations (4.3) and (4.6) may be used to write

$$F_x(k) = \frac{a^2}{2\pi} \Gamma(\frac{5}{3}) \sin(\frac{\pi}{3}) K_x (\text{grad } \bar{X})^2 \epsilon^{-1/3} k^{-5/3}. \quad (4.12)$$

The critical part of this study is the determination of a stable form for $F_x(k)$; that is, a form which lends itself to evaluation from experimental data, but with minimal uncertainty. Equation (4.12) is not well-suited for this purpose since the parameters involved range over several orders of magnitude. Atlas, Hardy and Naito [2] have introduced the outer scale explicitly. In the present instance this would be by⁷

$$C_x^2 = a^2 L_x^{4/3} (\text{grad } \bar{X})^2, \quad (4.13)$$

and

$$F_x(k) = \frac{a^2}{2\pi} \Gamma\left(\frac{5}{3}\right) \sin\left(\frac{\pi}{3}\right) L_x^{4/3} (\text{grad } \bar{X})^2 k^{-5/3}. \quad (4.14)$$

Since stable stratification will be presumed for the considerations here, $a^2 = a^2(R_i)$. Data presented by Lumley and Panofsky [33, p. 204], adapted from Tatarski [61], indicate that a^2 decreases with increasing R_i . Tatarski [60] has concluded from a somewhat sparse data sample that

$$a^2 = 0.1/R_i \quad (4.15)$$

for R_i greater than about 0.05. The uncertainty in $L_x^{4/3}$ is one order of magnitude. The range of $|\text{grad } \bar{X}|$ is of the order of a factor of ten. Although it would seem that these uncertainties would be resolved by evaluation from experimental data, the scale of observing is usually too

⁷ This presumes equality of the diffusivities, K_x and K_m .

large for an appropriate determination of gradients and Richardson's number.

Again following Atlas, Hardy and Naito [2], since the object here is to evaluate the refractive index spectrum in the presence of clear-air turbulence, it is appropriate to introduce some measure of the mechanical energy. The one-dimensional mechanical energy spectrum in the inertial subrange is

$$E(k) = c \frac{\Gamma(\frac{5}{3}) \sin(\frac{\pi}{3})}{2\pi} \epsilon^{2/3} k^{-5/3}, \quad (4.16)$$

where c is a numerical constant. If the spectral density is defined by

$$\int_0^{\infty} E(k) dk = \overline{(u')^2}, \quad (4.17)$$

where u' is the longitudinal velocity fluctuation, then $c \doteq 4$ for $k = 2\pi/\lambda$ (Panofsky and Pasquill [39]). The constant for transverse components is larger by a factor of $4/3$.

Equation (4.16) is presumed to be valid only for lag separations, $l_m \ll r \ll L_m$, where (Tatarski [61])

$$l_m = \left\{ (15c\nu)^3 / \epsilon \right\}^{1/4} \quad (4.18)$$

and L_m differs from L_x by the replacement $K_x \rightarrow K_m$. That is, the outer scales for momentum and the X-quantity are related by

$$L_m / L_x = s_*^{1/2} \quad (4.19)$$

where s_* is the turbulent Schmidt number.

A more critical factor in the consideration of overlapping of the inertial subranges are the microscales. For temperature and momentum the microscales are related by

$$\frac{l_m}{l_T} = \sigma^{3/4}. \quad (4.20)$$

The Prandtl number is 0.7 for the atmosphere so that

$$l_T \doteq 1.31 l_m \quad (4.21)$$

Tatarski's microscale for momentum is [eq. (4.18)]

$$l_{m1} = 4.54 \left(\frac{\nu^3}{\epsilon} \right)^{1/4} \quad (4.22)$$

This is not substantially different from the estimate by Megaw [37] used by Atlas, Hardy and Naito [2]. Pao [40] has indicated that the microscale for the convective subrange for arbitrary Schmidt number is

$$l_x = \left(\frac{\nu_x^3}{\epsilon} \right)^{1/4}. \quad (4.23)$$

This differs from Tatarski's estimate by a factor $2.28a^{3/2}$. For small Richardson number, this is of the order of 3.8. Thus, using either Tatarski's or Pao's results

$$l_x = l_m s^{-3/4}, \quad (4.24)$$

where s is the Schmidt number. As determined by Atlas, Hardy and Naito [2], the microscale in clear-air turbulence (in the jet stream

vicinity) is of the order of a few millimeters. According to their classification, moderate clear-air turbulence corresponds to an energy dissipation rate of the order of $100 \text{ cm}^2 \text{ sec}^{-3}$. Temperature spectra taken by Lowenberg, Bostick and McCullough [32] show definite slope discontinuities at a scale of the order of 10 cm .⁸ If the average value of ϵ (100 m) given by Wilkins [62] is used, then (using 4.23)

$$\ell_T = \left\{ \frac{(0.19 \text{ cm}^2 \text{ sec}^{-1})^3}{2 \text{ cm}^2 \text{ sec}^{-3}} \right\}^{1/4} = 2.4 \text{ mm.}$$

Using Tatarski's estimate, $\ell_T \approx 1 \text{ cm}$. Since the "true" mean wind speed estimate for the Lowenberg et al. data would likely yield still a greater scale, estimates of the refractive index spectrum corresponding to lengths ($\lambda/2$) may not be in the inertial subrange for wavelengths of a few centimeters (for a backscattering evaluation). It is of interest to note that the Lowenberg et al. data show an average slope of -1.65 at low frequencies. This is in excellent agreement with the "-5/3" law. At higher frequencies they found an average slope of -2.99. Here it will be presumed that the inertial and convective subranges coincide and the spectra determinations required may be evaluated in the inertial subrange.

Equations (4.12) and (4.16) may be combined to write

⁸ There is some uncertainty here because the observations were taken on a tower, but mean wind speed estimates are at the surface.

$$F_x(k) = \frac{a^2 K_x (\text{grad } \bar{X})^2}{c K_m (\partial \bar{u}_i / \partial X_j)^2} E(k). \quad (4.25)$$

Following Tatarski [61], it is presumed that the mean index of refraction depends only on the vertical coordinate so that

$$|\text{grad } \bar{N}| = \frac{10^{-6} A p}{T^2} \left\{ 1 + \frac{3.24 B q}{T} \right\} \left\{ \frac{\partial \theta}{\partial z} - \frac{1.62 B}{1 + \frac{3.24 B q}{T}} \frac{\partial q}{\partial z} \right\}, \quad (4.26)$$

where p is pressure, T is absolute temperature, q is specific humidity, θ is potential temperature, and A and B are empirical constants. When p is expressed in millibars, $A \cong 79$ and $B \cong 4800$. Typical values of q in the jet stream vicinity are of the order of $1-300 \text{ gm}/10^6 \text{ gm}$ [6]. Thus, the term $(3.24 Bq/T)$ may be neglected when compared to unity. However, as suggested by Figs. 12 and 13, after Briggs and Roach [6], the moisture term is not necessarily negligible, as taken by Atlas, Hardy and Naito [2]. Figure 13 illustrates the intrusion of dry stratospheric air into the baroclinic zone below and on the cyclonic side of the jet core. Thus, for

$$|\text{grad } \bar{N}| = \frac{10^{-6} A p}{T^2} \left\{ \frac{\partial \theta}{\partial z} - 7800 \frac{\partial q}{\partial z} \right\}, \quad (4.27)$$

the moisture gradient on the upper boundary will serve to decrease the radar return. The moisture contribution on the upper boundary is of the order of 40% in this instance, but is somewhat less elsewhere. Since this was apparently the most pronounced case during the Briggs and Roach investigation, moisture considerations are also to be omitted here. It

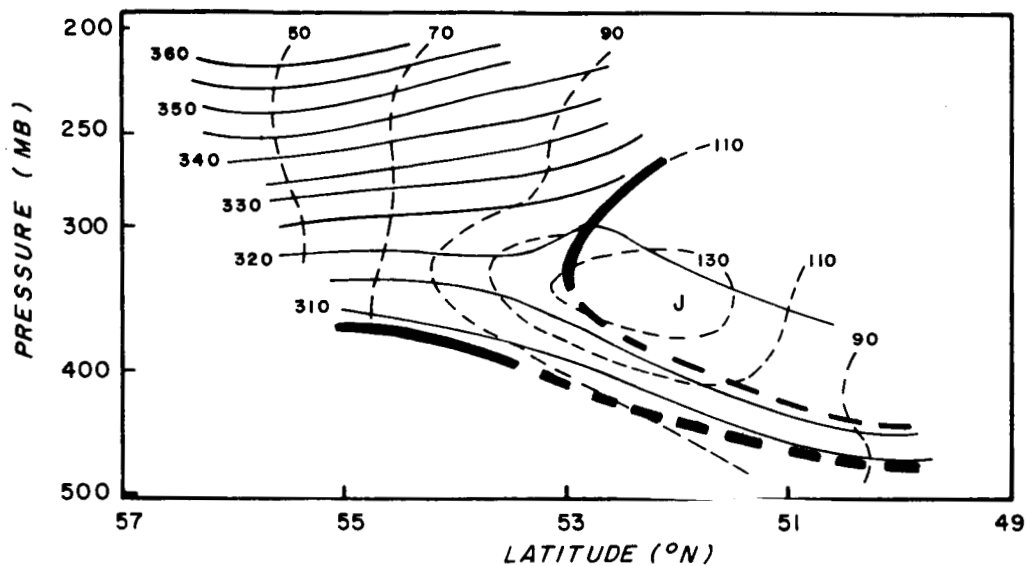


FIG.12. FLIGHT CROSS SECTION ANALYSIS.

*Dashed light lines are isotachs (KT).
Solid lines denote potential temperature.
(after Briggs and Roach [6])*

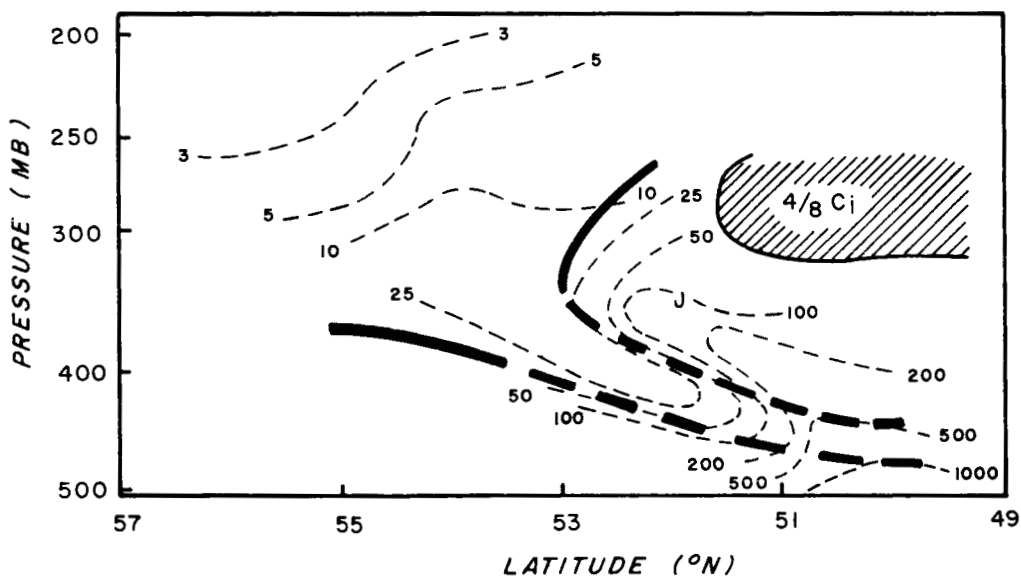


FIG.13. FLIGHT CROSS SECTION ANALYSIS.

*Dashed light lines are isopleths of mixing ratio
(GM/10 GM) (after Briggs and Roach [6])*

should be noted that an enhanced radar return will result from the moisture distribution everywhere but the upper boundary of the jet stream front.

If it is now assumed that K_x may be represented adequately by the diffusivity for heat, K_h , then

$$F_N(k) = \frac{a^2 |R_f|}{g c} f(p, T) \left| \frac{\partial \theta}{\partial z} \right| E(k), \quad (4.28)$$

where

$$f(p, T) = T^{-3} \{79 \times 10^{-6} p(\text{mb})\}^2. \quad (4.29)$$

The approximation $T \cong \theta$ has been used.

Since there are no measurements of $E(k)$ for k values of interest, it is convenient to rewrite (4.28) as

$$F_N(k) = \frac{a^2 |R_f|}{g c} f(p, T) \left(\frac{k_1}{k}\right)^{5/3} E(k_1) \left| \frac{\partial \theta}{\partial z} \right|. \quad (4.30)$$

The function $f(p, T)$ may be evaluated with negligible error, a reliable value of c (≈ 4) is available, and $E(k_1)$ has been measured by Reiter and Burns [52] for various intensities of clear-air turbulence. Atlas, Hardy and Naito [2] used $E\left(\frac{2\pi}{100\text{m}}\right) = 3.2 \times 10^6 \text{ cm}^2 \text{ sec}^{-2} / \text{rad sec}^{-1}$ as indicative of moderate turbulence.^{9, 10} This will be used here along with the assumption

⁹ As indicated by MacCready's results [35], the extrapolation to the centimeter range is probably not in serious error.

¹⁰ This is slightly larger than suggested by MacCready [36] and is of the order of surface boundary layer values [62].

$$a^2 |R_f| = 0.1. \quad (4.31)$$

Since the atmosphere is polytropic to a large extent, $f(p, T)$ is essentially a function of pressure only. A suggested model is shown in Fig. 14. An essentially precise evaluation of $f(p, T)$ can be made for any realization; however, for purposes of estimation, the few per cent deviation from Fig. 14 at any given pressure is of no consequence. The point corresponding to $p = 300$ mb, $T = 223$ K, after Atlas et al., is also shown. The variation is significant enough to warrant its accounting.

Equation (4.30) can now be written as

$$F_N(k) = 3.7 \times 10^{-4} f(p, T) \left| \frac{\partial \theta}{\partial z} \right| k^{-5/3} \left(\frac{1}{\frac{\text{radians}}{\text{cm}}} \right). \quad (4.32)$$

The principal assumptions used in obtaining (4.32) were: (1) the existence of an inertial subrange, (2) the energy level and extrapolation, and (3) the relation $a^2 |R_f| = 0.1$. The latter is suspect, but since a^2 is a monotonically decreasing function of the Richardson number, the assumed functional relationship should be indicative of its behavior. As suggested by (4.30), the refractive index spectrum is determined by both the thermal (and moisture in the general case) configuration and the intensity of turbulence.

Before calculating the spectrum for some sample profiles, it may be worthwhile pointing out that eq. (4.32) presumes that moderate turbulence exists. This, of course, implies a vector wind shear, but for $R_i > 0.05$,

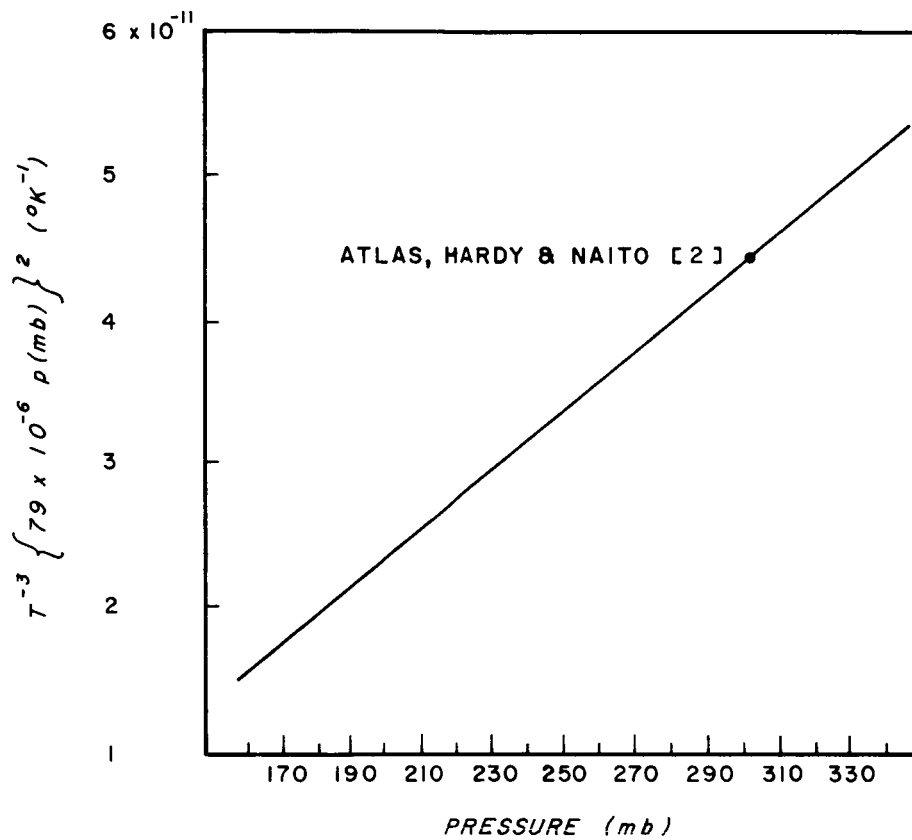


FIG. 14. ASSUMED $f(p, T)$ RELATIONSHIP AS A FUNCTION OF PRESSURE

consistency is no problem. However, it should be mentioned that more straightforward approaches must cope with consistency. Atlas, Hardy and Naito [2] have evaluated a^2 for $R_i = 0.05$ and have stipulated p , T and $\partial\theta/\partial z$. This could be used to imply a wind shear:

$$\left(\frac{\partial \bar{u}_i}{\partial x_j}\right)^2 = \frac{g}{\theta} \frac{\partial\theta/\partial z}{R_i}. \quad (4.33)$$

For their choice of parameters, $\left|\left(\frac{\partial \bar{u}_i}{\partial x_j}\right)\right| \sim 8 \times 10^{-2} \text{ sec}^{-1}$. While this is quite large, it may be physically realizable. Stinson, Weinstein and Reiter [59] have reported values as high as $5 \times 10^{-2} \text{ sec}^{-1}$. However, the concern here is that the choices of ϵ and L_o also imply a wind gradient:

$$\left(\frac{\partial \bar{u}_i}{\partial x_j}\right)^2 = \frac{\epsilon^{2/3}}{L_o^{4/3}}. \quad (4.34)$$

Using the minimum L_o (10^4 cm) and $\epsilon = 400 \text{ cm}^2 \text{ sec}^{-3}$,

$$\left|\frac{\partial \bar{u}_i}{\partial x_j}\right| = 1.65 \times 10^{-2} \text{ sec}^{-1}.$$

These shear estimates differ by a factor of five.

The illustration of eq. (4.32) chosen here is taken from data given by Endlich [20] and shown in Fig. 15. The potential temperature gradients along the lines marked A and B were estimated; smoothed profiles are shown in Fig. 16.

The results shown in Fig. 17 are expressed in terms of the coefficient C_N^2 for ease in comparison with other results (cf. eq. 4.11). Atlas, Hardy and Naito [2] found that an appropriate range for C_N^2 was $10^{-16} \neq 10^{-14} \text{ cm}^{-2/3}$.

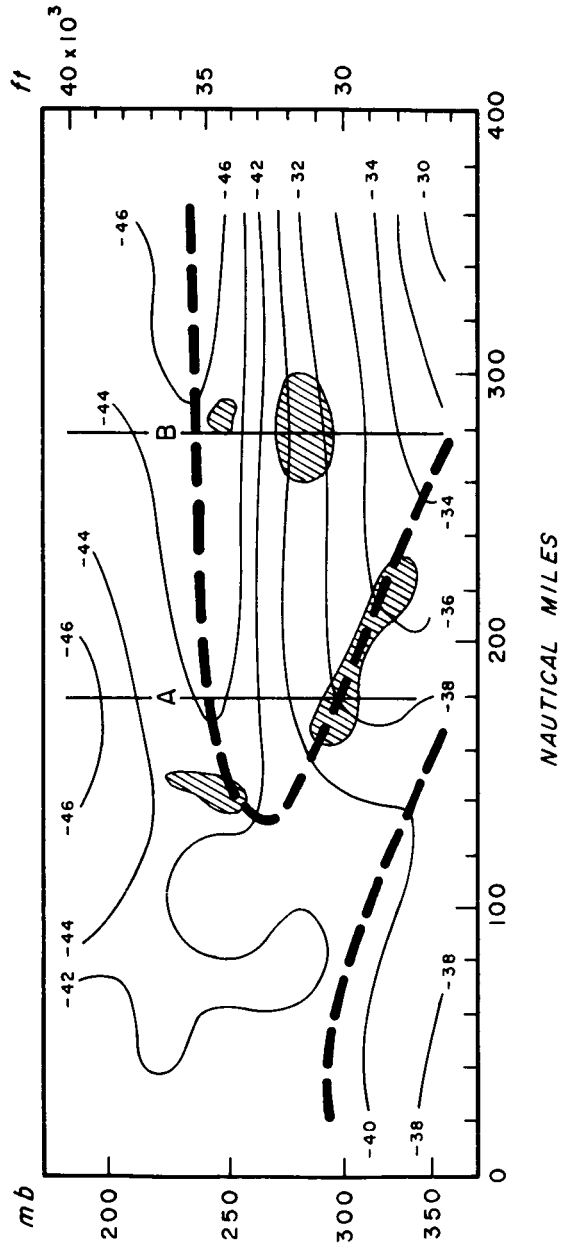


FIG. 15. VERTICAL CROSS SECTION OF TEMPERATURE ($^{\circ}\text{C}$) FROM THE LOW-PRESSURE (LEFT) TO HIGH PRESSURE (RIGHT) SIDE OF THE JET STREAM. REGIONS OF MODERATE TURBULENCE SHADED. SOUNDINGS AT A AND B USED FOR SAMPLE REFRACTIVE INDEX SPECTRA COMPUTATIONS. (after Endlich [20J])

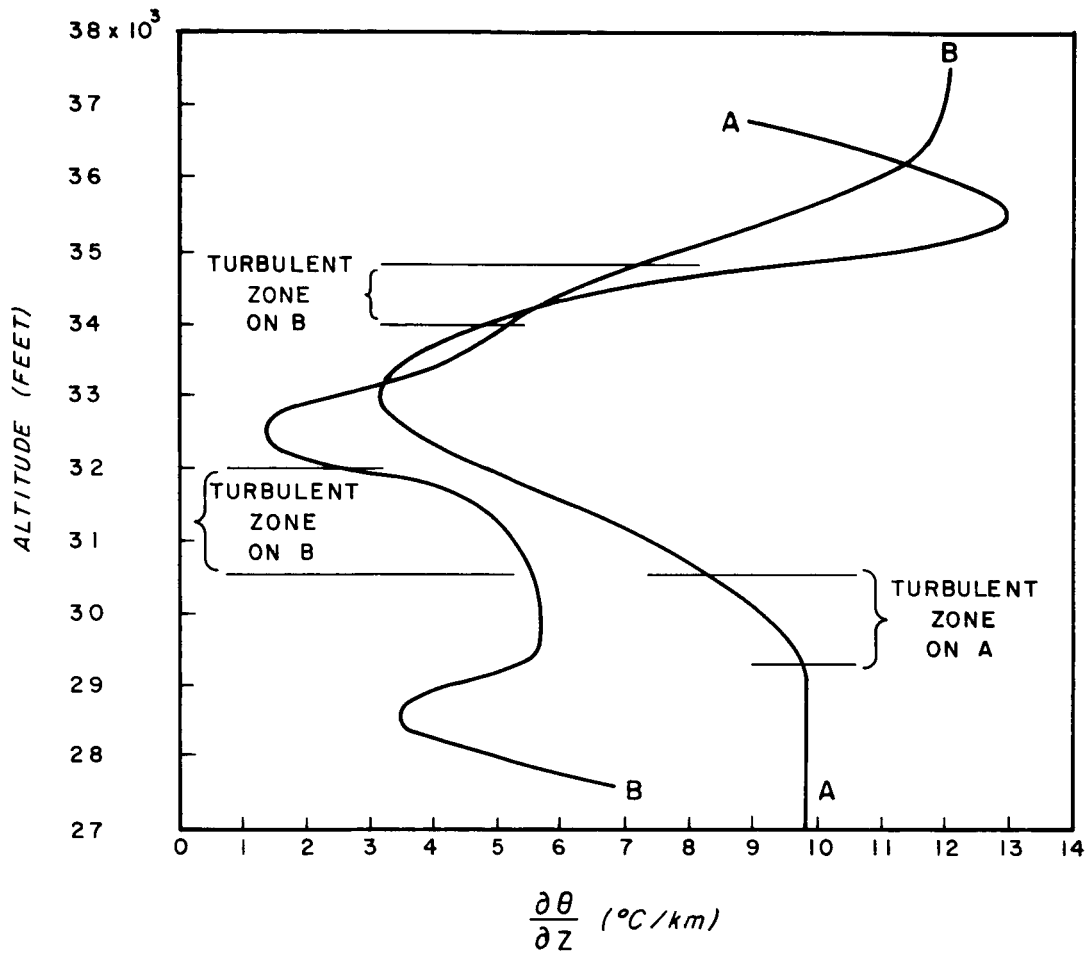


FIG. 16. SMOOTHED PROFILES OF POTENTIAL TEMPERATURE GRADIENT

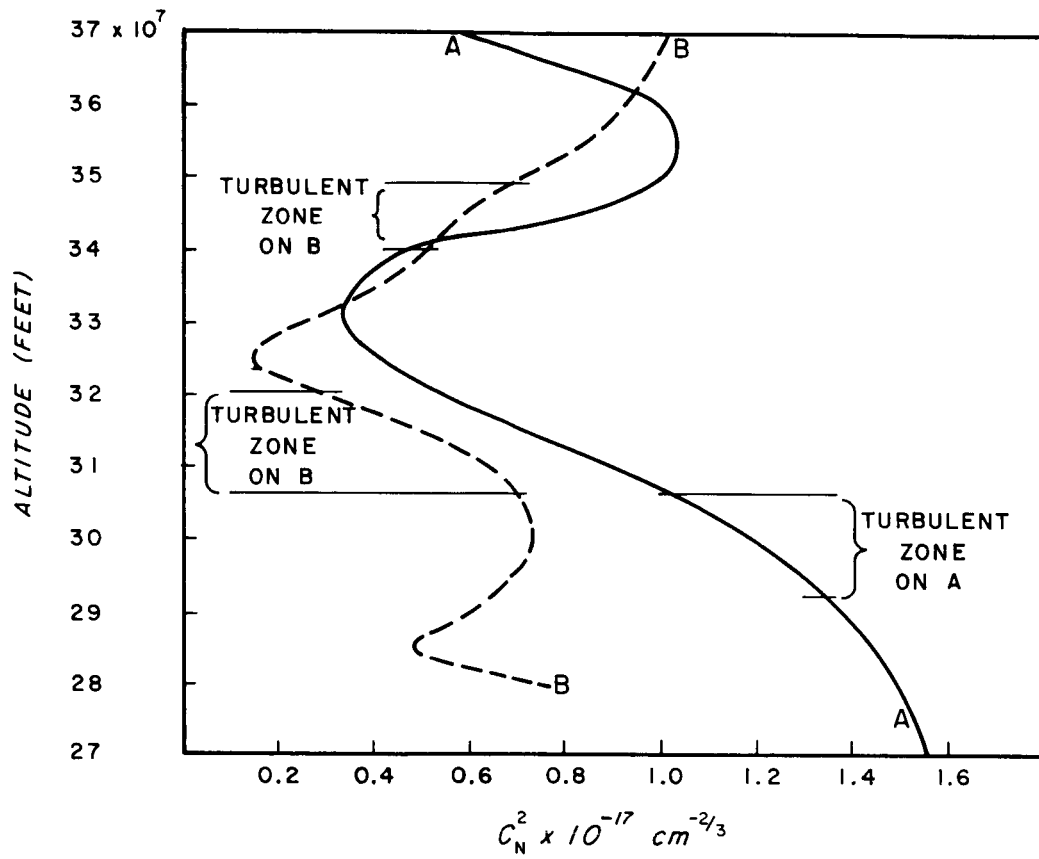


FIG. 17. FLUCTUATION AMPLITUDE COEFFICIENT C_N^2 AS A FUNCTION OF ALTITUDE FOR SOUNDINGS A AND B (see fig. 17)

The results here are one to three orders of magnitude less. Using an outer scale of 600 m, the p , T and $\partial\theta/\partial z$ values by Atlas et al., and the relation $a^2 R_i = 10^{-1}$, a value $R_i \approx 0.06$ is obtained for $C_N^2 = 10^{-17}$. Smaller values of C_N^2 yield larger R_i . The disparity can be partially explained by noting that for a wind shear of $1.65 \times 10^{-2} \text{ sec}^{-1}$ and a lapse rate of $10^{-4} \text{ }^\circ\text{C/cm}$, Atlas et al. should have evaluated $a^2 \Big|_{R_i=1.2} \approx 1/12$. Thus, their results may be overestimated by a factor of 24. This explains a major portion of the discrepancy.

Again, Fig. 17 has been constructed on the tacit assumption that moderate CAT is occurring everywhere. When energy levels (which enter the calculation of C_N^2 linearly) are reduced to non-turbulent levels (Wilkins [62]), C_N^2 will be orders of magnitude smaller than indicated in Fig. 17.

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