

General Disclaimer

One or more of the Following Statements may affect this Document

- This document has been reproduced from the best copy furnished by the organizational source. It is being released in the interest of making available as much information as possible.
- This document may contain data, which exceeds the sheet parameters. It was furnished in this condition by the organizational source and is the best copy available.
- This document may contain tone-on-tone or color graphs, charts and/or pictures, which have been reproduced in black and white.
- This document is paginated as submitted by the original source.
- Portions of this document are not fully legible due to the historical nature of some of the material. However, it is the best reproduction available from the original submission.

BOEING

GPO PRICE \$ _____

CFSTI PRICE(S) \$ _____

Hard copy (HC) 300

Microfiche (MF) _____

ff 653 July 65

FACILITY FORM 602

N 68-26806
(ACCESSION NUMBER) (THRU)
22
(PAGES) (CODE)
NASA-CR-61810
(NASA CR OR TMX OR AD NUMBER) (CATEGORY)
15



SPACE DIVISION

LAUNCH SYSTEMS BRANCH

THE **BOEING** COMPANY
AERO-SPACE DIVISION
SATURN BOOSTER BRANCH

DOCUMENT NO. T5-6556-18

VOLUME _____ OF _____

TITLE Evaluation of 3M's Y-9050 Tape For Sealing Openings at the
Intertank - to - Y - Ring Joint

MODEL NO. Sat V/S-IC CONTRACT NO. NAS8-5608

ISSUE NO M-14 ISSUED TO R. L. Murphy
National Aeronautics
And Space Administration

PREPARED BY	<u>R. O. Koch</u>	<u>12-1-67</u>
	<u>R. O. Koch</u> 5-7996	
SUPERVISED BY	<u>E. L. Clark</u>	<u>2-24-68</u>
	<u>E. L. Clark</u> 5-7996	
APPROVED BY	<u>W. W. Davis</u>	<u>3-1-68</u>
	<u>W. W. Davis</u> 5-7900	
APPROVED BY	_____	_____
APPROVED BY	_____	_____
APPROVED BY	_____	_____

REV. SYM. _____

S
PRECEDING PAGE BLANK NOT FILMED.

ABSTRACT

Tests were conducted to determine the suitability of 3M's Y-9050 Tape for sealing openings at the Intertank-to-Y-Ring joint. Peel tests were conducted over a temperature range of 75° to -250°F. Also the leading edges of the tape were subjected to a blast of air (@ ~ 75°F) to simulate the air flow encountered in flight. The performance of the tape was considered sufficient to meet Design Engineering requirements.

Key Words

Intertank-to-Y-Ring Joint

Y-9050 Tape or 60B32045 Type 14 Tape

Adhesion Test

Air stream Test

TABLE OF CONTENTS

REF. NO.		PAGE
	Distribution	ii
	Change Record Page	iii
	Revisions Page	iv
	Abstracts and Key Words	v
	Table of Contents	vi
	List of Illustrations	vii
1.0	OBJECT	1
2.0	BACKGROUND	1
3.0	CONCLUSIONS	1
4.0	TEST PROCEDURES	1
5.0	TEST RESULTS	3

LIST OF ILLUSTRATIONS

Figure No.	Title	Page
1	Aluminum Backing Plate For Peel Test	5
2	Aluminum Backing Plate With Tape Applied	6
3	Typical Test Set-UP For Peel Test	7
4	Test Fixture For Air Stream Test - Before Application of Tape	8
5	Air Stream Test Fixture With Tape Applied	9
6	Peel Strength vs. Temperature of 3M's Y-9050 Tape	10
Table No.		
I	Peel Test Data of 3M's Y-9050 Tape	11

1.0 OBJECT

To determine the suitability of MSFC-SPEC-60B32045 type 14 tape (3M's Y-9050) to seal openings at the intertank-to-Y-ring joints.

2.0 BACKGROUND

The above mentioned openings require sealing to prevent the entrance of air, thus preventing excess frost build-up, by freezing of condensate, during "lox fill".

To determine the suitability of the tape for this purpose the peel adhesion was determined over the expected temperature service range (ambient to -250°F, nominal). In addition, the tape was subjected to an air stream velocity of 30 psia to indicate whether or not it would blow off during flight.

3.0 CONCLUSIONS

The peel adhesion of Y-9050 tape over the expected temperature exposure range and the ability of the tape to resist the expected air stream velocity was considered sufficient to meet design requirements.

4.0 TEST PROCEDURES

4.1 PEEL TEST PROCEDURE

4.1.1 Backing Plates

7075-T6 aluminum plates (Figures 1 & 2), were used as the adherends from which the Y-9050 tape was peeled. The initial surface preparation of the plates consisted of: iridizing per BAC 5726 then overcoating with MIL-P-8585 zinc chromate primer applied by aerosol spray application per BAC 5776. Because a limited number of these plates were available each was subjected to repeated use. Some of the tests resulted in detachment of the zinc chromate primer; in such cases it was removed with a ketone solvent and reapplied, either by brushing or spraying as noted in the attached table, and given an accelerated cure per BAC 5776.

4.1.2 Application of Tape to Plates

Three inch wide strips of tape were applied to the plates by two different methods: with a roller per ASTM D1000, except the roller weight was only 8 pounds per 3 inches of width; and with hand pressure. To minimize procedure variations all specimens were prepared by the same technician.

4.1.3 Test Temperature

The various test temperatures, noted in Table I, were attained by adding the appropriate coolant, below, to the test fixture, , shown in Figure 3. The temperature was monitored with a thermocouple attached to the backing plate as shown in Figure 2.

Coolants Used to Obtain Low Temperature

Nominal Stabilization Temp.	Coolant
-100°F	Methanol + Dry Ice
-200°F	Ethanol + LN ₂
-320°F	Liquid Nitrogen

 Test Fixture: Made from 1/2 inch thick 321 stainless steel. Dimensions: 14 (height) x 12 x 6 inches.

Low temperatures were attained with the coolant which stabilized on the low side, at a temperature nearest the desired test temperature. Tests conducted at low temperatures above the stabilization temperature were performed as the temperature dropped. This resulted in a temperature range as shown in Table I.

Ambient tests were performed by filling the reservoir with ambient air.

4.1.4 Testing

The specimen was set up as shown in Figure 3. After reaching the desired temperature the tape was peeled from the backing plate at 180° at a rate of 12 inches per minute (ref. ASTM D1000). The tensile test machine used was "Baldwin 20K NASA #045482".

4.2 AIR STREAM TEST PROCEDURE

4.2.1 General

The following general procedure was used to determine the ability of Y-9050 tape to withstand the maximum expected air stream velocity that it will encounter during flight. The various air pressures referred to in the procedure were as specified by the organization that requested this test .

 Coordination Sheet No. I&HSG-M-87, from P. E. Oster to A. Paxhia, dated 2-7-66, Subject: Request for Test Support.

4.2.2 Application of Air Stream

Openings or gaps, approximately four square inches in area, in a test fixture configured to simulate the propellant tank to Y-ring junction (figure 4), were sealed with strips of Y-9050 tape as shown in Figure 5. An air stream (30 psia) was applied perpendicular to the leading edge of the tape strips for a maximum period of 150 seconds. Four tests were conducted, two in each direction (up & down), to simulate the junction at each of the propellant tanks (lox and fuel).

4.2.3 Internal Pressure

To further simulate flight conditions a slight positive pressure differential was applied, simultaneously with the air stream test, to the inboard side of the fixture. This was accomplished by fabricating a balloon type seal, out of polyethylene, over an area to include the taped openings. Air pressure was applied, in the balloon seal, in an attempt to obtain a pressure differential of 1.5 psi; but due to uncontrollable leaks a maximum of only 0.4 psi was attained as indicated by a water filled manometer. Due to a tight schedule and to Design's opinion that attaining the 1.5 psi differential did not warrant the extra time and effort necessary to prepare a leak proof seal the test was conducted at the 0.4 psi (max.) differential.

5.0 RESULTS

5.1 PRESENTATION OF DATA

Individual peel test values along with averages for specimens prepared and tested under similar conditions are presented in Table I, attached. Figure 6 shows a plot of the average values at various temperatures [?].

The following discussion considers general peel test results based on Figure 6 ; but details the air stream test results.

5.2 PEEL TEST RESULTS

From Figure 6 it is apparent that the adhesion of Y-9050 tape increased from a nominal 5#/3 inch of width at ambient room temperature to a corresponding high of 20#/3 inches in the proximity of -100°F, then dropped rapidly to a nominal low of 2#/3 inches around -250°F.

3 > Although symbols on Figure 6 distinguishes between the various types of specimen preparation procedures, the average curve was drawn without consideration of this variable. This was done due to limited data for each type of preparation. However, the curve is considered valid because of the closeness of results of specimens tested at comparable temperatures.

5.2 PEEL TEST RESULTS (Continued)

The type of specimen preparation, that is, how the primer and tape were applied made no real significant difference as indicated by peel results compared at a common test temperature.

At temperatures above -200°F the plane of failure was at the tape adhesive/zinc chromate primer interface. Below -200°F failure occurred at different planes; but mostly at the same plane mentioned above.

5.3 AIR STREAM TEST

Out of three of the four tests conducted, the leading edge of the tape began to peel back after withstanding the 30 psia air blast for a minimum of 110 seconds. During the remainder of the 150 second test period, peel in two of the tests progressed to the extent that the sealed opening was exposed. This occurred at a valley opening (See Figure 4) in both cases; also it occurred when the air stream was applied from either direction.

Although peel took place as mentioned above, in no case was the tape blown completely off.

5.4 SUMMARY

The performance of the tape was considered sufficient to meet design requirements since: (1) some degree of adhesion was retained at the temperature extreme of -250°F, and (2) it did not blow off during 150 second flight simulated air flow test.

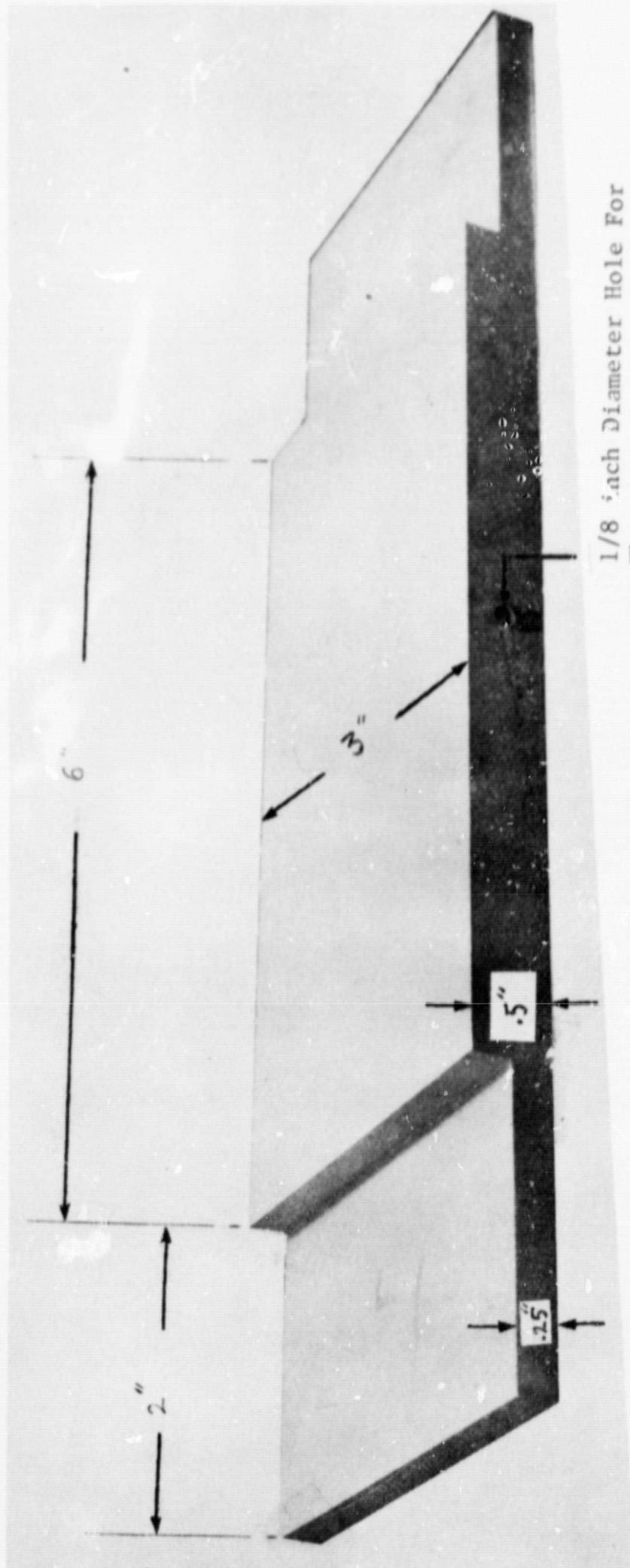


FIGURE 1

ALUMINUM BACKING PLATE FOR PEEL TEST

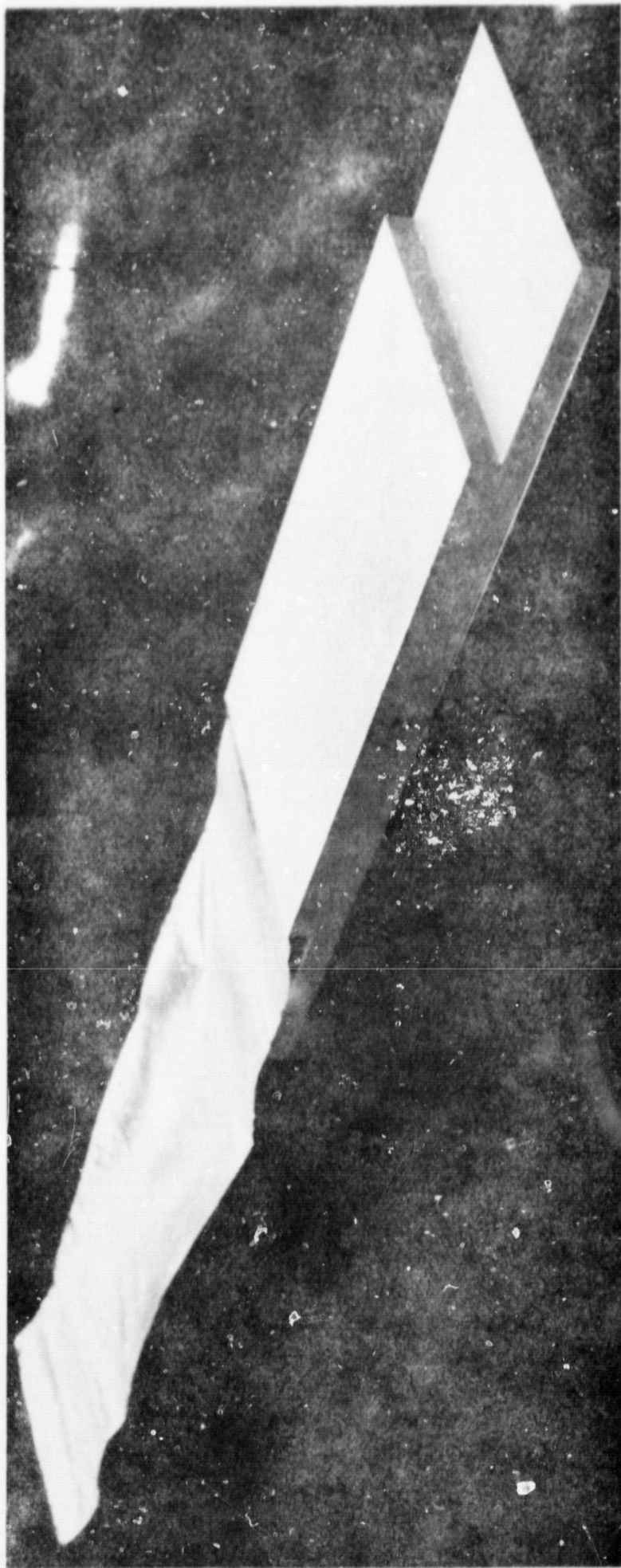


FIGURE 2
ALUMINUM BACKING PLATE WITH TAPE APPLIED

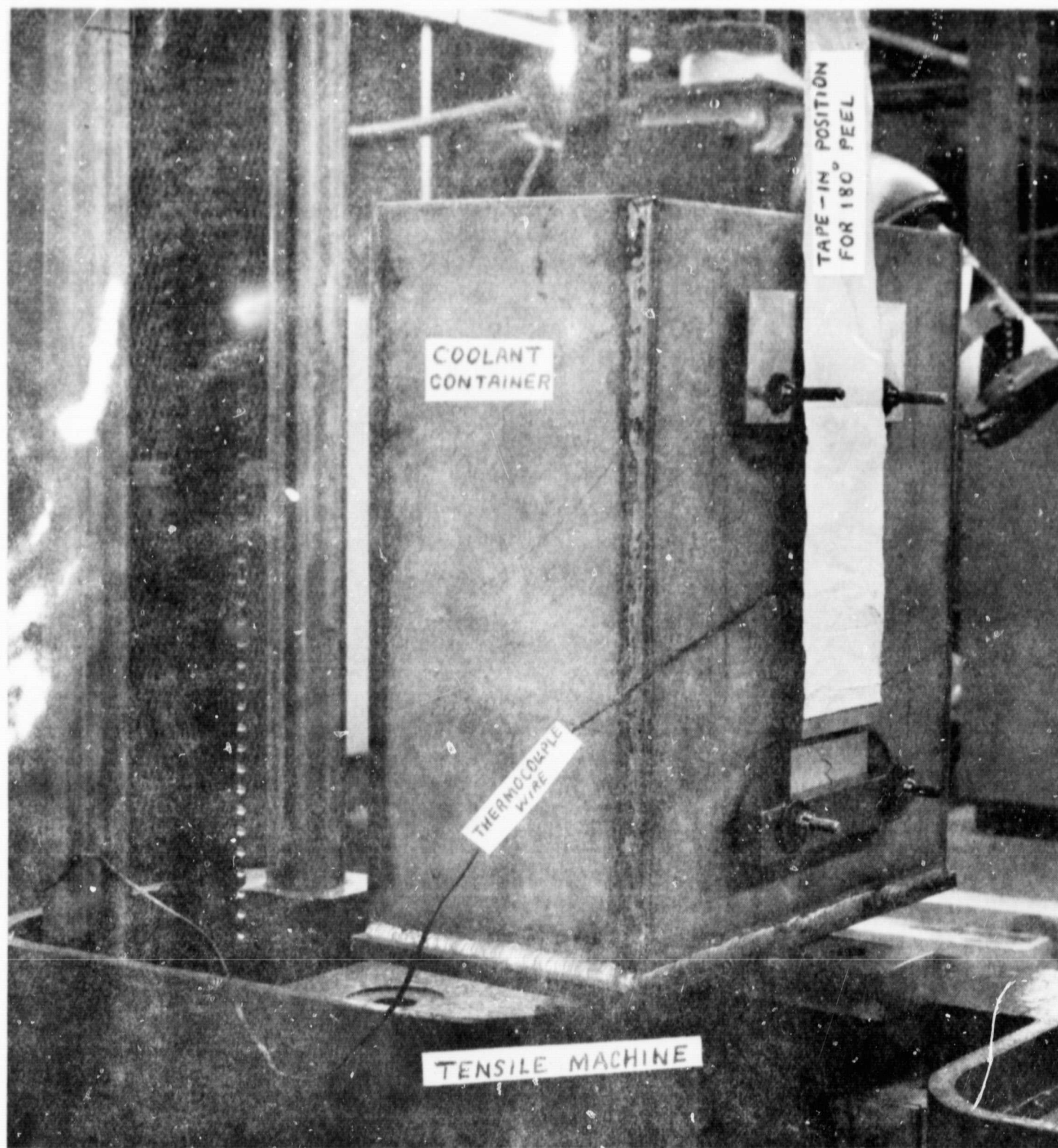


FIGURE 3
TYPICAL TEST SET-UP FOR PEEL TEST

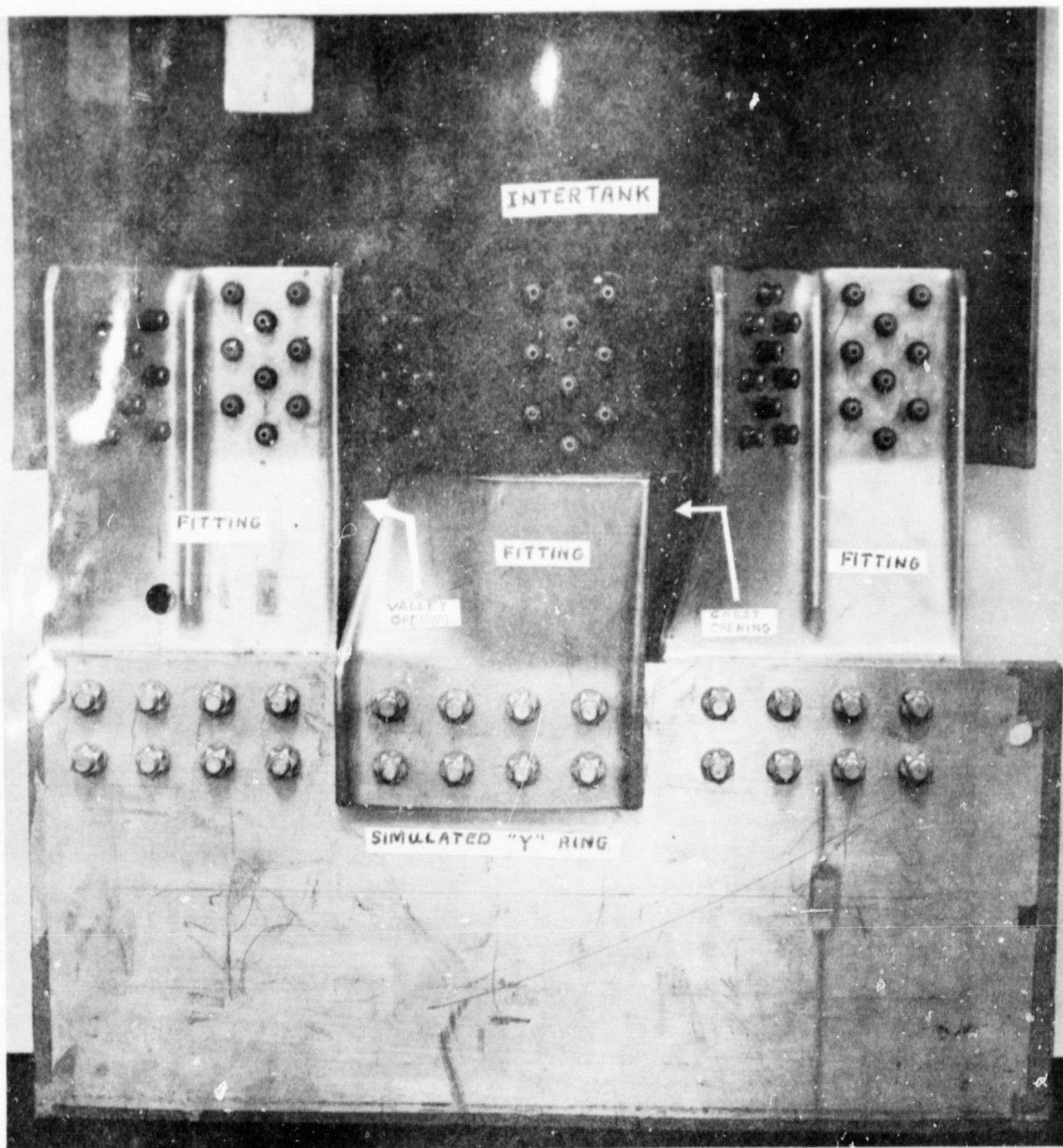


FIGURE 4

TEST FIXTURE FOR AIR STREAM TEST - BEFORE APPLICATION OF TAPE

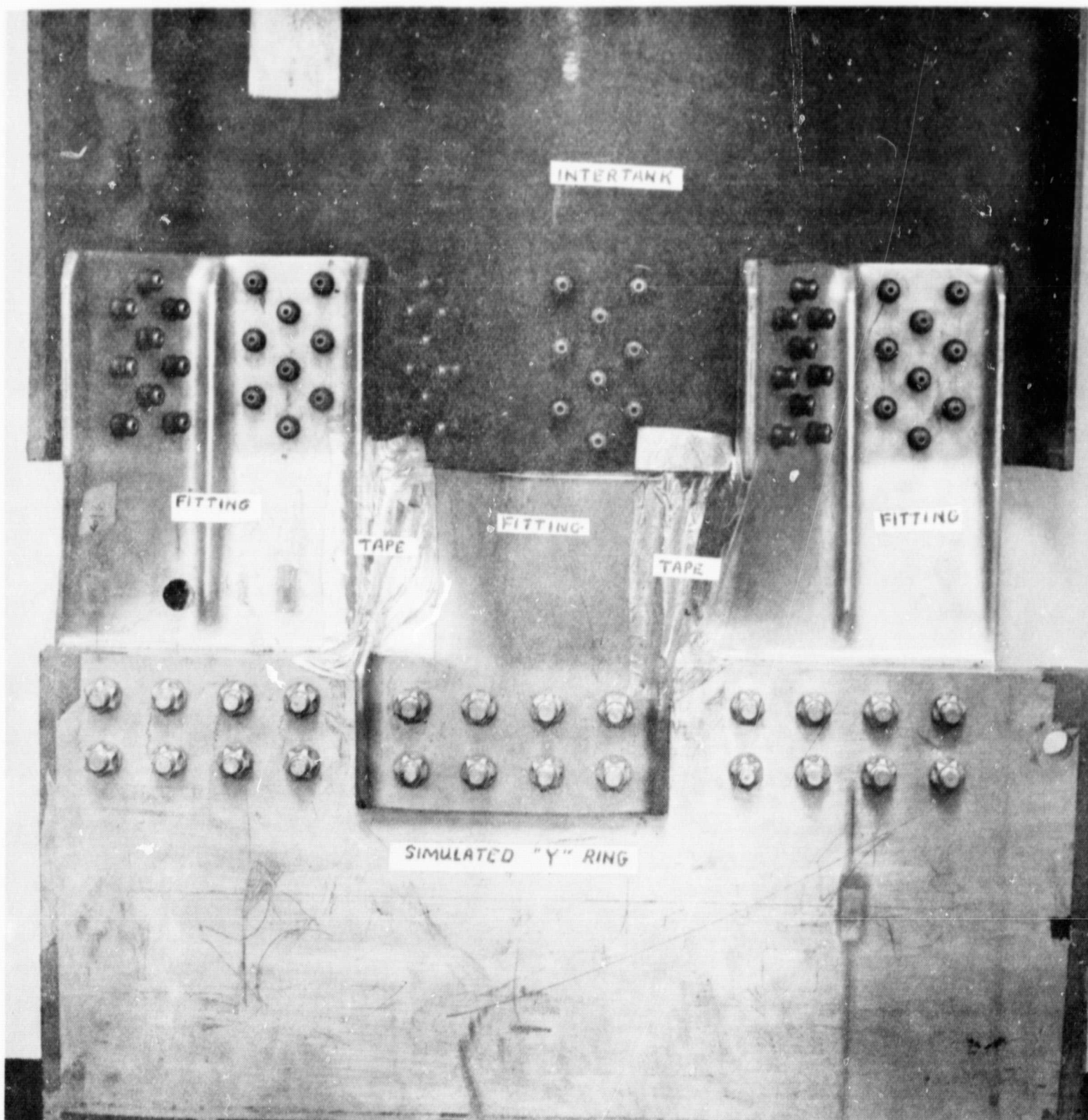
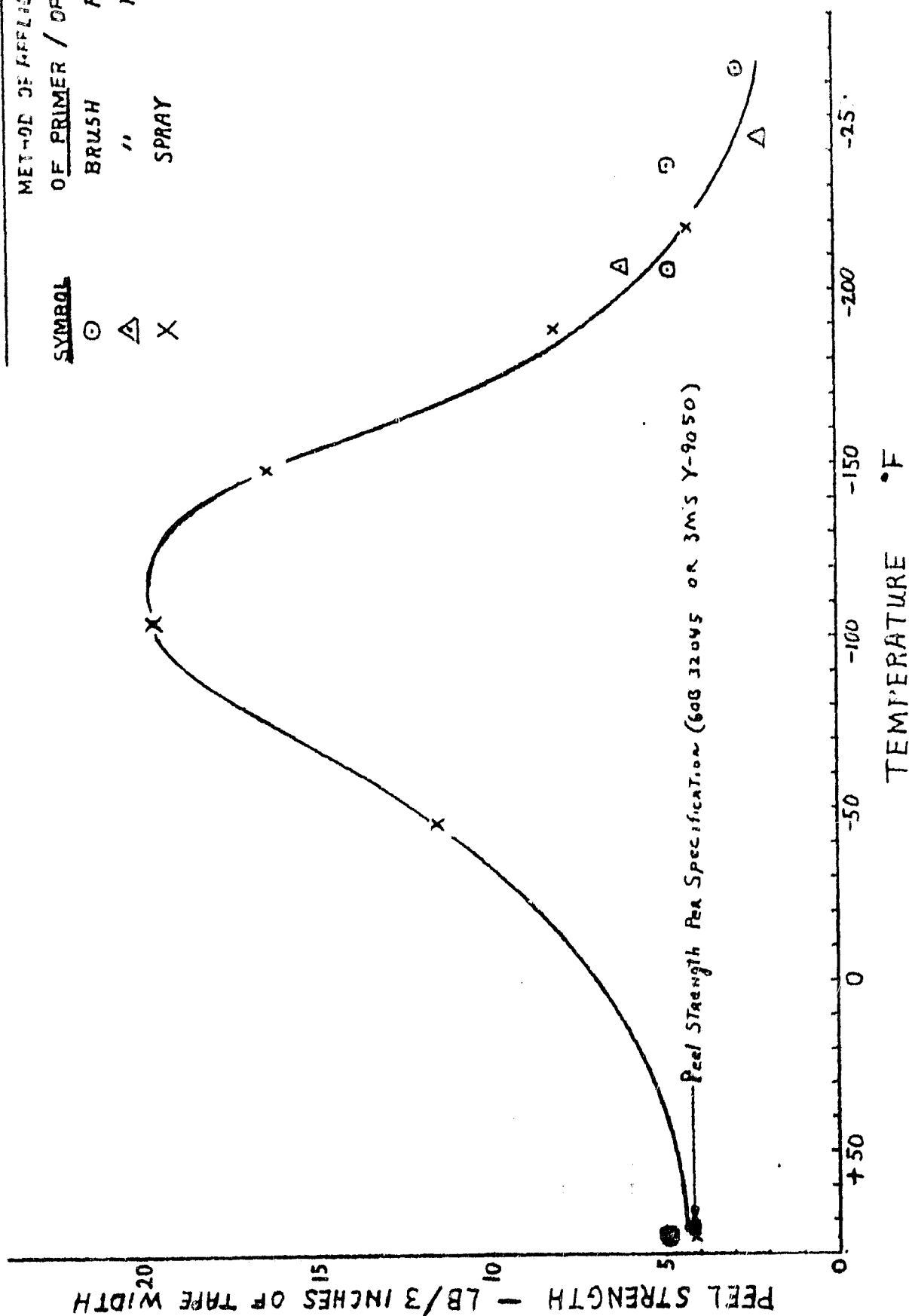


FIGURE 5

AIR STREAM TEST FIXTURE WITH TAPE APPLIED

LEGEND:

METHOD OF APPLICATION:		
OF PRIMER / OF TAPE		
SYMBOL	BRUSH	ROLLER
○	"	HAND
△	"	"
X	SPRAY	"



CALC			REVISED	DATE
CHECK				
APR				
APR				

Figure 6
PEEL STRENGTH VS. TEMPERATURE OF 3M'S
Y-9050 TAPE

THE DOENG COMPANY

FOLDOUT FROM 1

SPEC. NO.	TEST TEMP.		NOMINAL AVG	TEST VALUES - LB/3 INCHES OF WIDTH - READINGS TAKEN @ ½ INCH INTERVALS									
	RANGE			1	2	3	4	5	6	7	8	9	10
	°F												
1	75 ± 5°F			5.0	5.0	5.2	4.8	4.6	4.8	5.0	4.8	4.2	3.6
2				5.6	5.4	5.4	5.6	5.2	5.6	4.8	4.8	4.4	5.4
3				6.0	6.0	5.8	5.8	6.0	5.8	5.6	5.6	5.6	5.8
4				4.8	4.6	4.2	4.0	4.6	4.5	4.6	4.4	5.0	5.2
5				5.0	4.3	4.0	4.6	4.2	4.2	4.2	4.2	4.2	4.2
				(AVG. OF SPECIMENS 1-5)									
6				5.0	5.0	5.0	4.8	5.0	5.0	4.6	4.6	4.6	4.6
7				4.8	5.0	5.0	5.2	5.6	5.4	5.4	5.2	5.6	5.6
8				4.2	5.0	5.0	4.4	4.8	5.0	4.8	4.8	4.8	4.8
				(AVG. OF SPECIMENS 6-7)									
9				4.5	4.4	4.3	4.4	4.1	3.8	3.6	3.5	3.5	-
10				4.6	4.4	4.2	4.2	4.4	4.4	4.4	4.2	4.1	4.8
				(AVG. OF SPECIMENS 9 & 10)									
11	-100	-112	-106	19.0	17.0	20.8	20.0	19.0	18.6	17.0	15.2	15.4	-
12	-100	-120	-110	22.4	23.4	22.0	21.4	21.2	20.6	19.4	19.6	21.4	22.4
13	-100	-102	-101	22.0	21.4	20.8	20.8	21.6	19.8	17.0	17.2	17.6	17.2
(AVG. SPEC. 11 - 13)			-105										
14	-45	-50	-48	10.0	8.6	9.0	11.6	10.4	11.2	10.8	10.6	10.0	9.8
15	-42	-45	-44	13.0	12.8	13.0	13.5	13.8	12.4	12.8	11.8	11.0	10.8
(AVG. SPEC. 14 & 15)			-46										
16	-200	-210	-205	6.8	6.0	5.8	5.4	6.4	6.0	6.0	4.0	3.8	3.4
17	-205	-209	-207	3.6	4.0	4.2	5.0	6.0	5.2	3.6	3.2	3.8	4.2
(AVG. SPEC. 16 & 17)			-206										
18	-233	-238	-236	4.0	5.0	4.8	5.4	6.0	4.4	4.2	4.4	4.6	5.0

Calc.			REVISED	DATE	PEEL TEST DATA OF 3M'S Y-9050 TAPE	PAGE 1 of 2	DATA S
Trac							
Chk						TABLE I	
Appr.							
Appr.					THE BOEING COMPANY SEATTLE 24, WASHINGTON		

FOLDOUT FRAME 2

- READINGS				PLAIN OF FAILURE			METHOD OF APPLYING ZINC CHROMATE PRIMER		METHOD OF APPLYING TAPE		NOTES:
8	9	10	AVG	1/N	2/N	3/N	BRUSH	SPRAY	8 LB ROLLER	HAND PRESSURE	
.8	4.2	3.6	4.7	1/100			X		X		
.8	4.4	5.4	5.2	1/100			X		X		
.6	5.6	5.8	5.8	1/100			X		X		
.4	5.0	5.2	4.6	1/100			X		X		
.2	4.2	4.2	4.4	1/100			X		X		
			4.9								
.6	4.6	4.6	4.8	1/100			X			X	
.2	5.6	5.6	5.3	1/100			X			X	
.8	4.8	4.8	4.8	1/100			X			X	
			5.0								
.5	3.5	-	4.0	1/100				X		X	
.2	4.1	4.8	4.4	1/100				X		X	
			4.2								
.2	15.4	-	18.0	1/100				X		X	
.6	21.4	22.4	21.4	1/100				X		X	
.2	17.6	17.2	19.5	1/100				X		X	
			19.6								
.6	10.0	9.8	10.2	1/100				X		X	
.8	11.0	10.8	12.9	1/100				X		X	
			11.5								
4.0	3.8	3.4	5.2	1/90		3/10	X		X		
3.2	3.8	4.2	4.3	1/90		3/10	X		X		
			4.8								
4.4	4.6	5.0	4.8	1/100			X		X		

DATA SHEET

YING

HAND
MEASURE

NOTES:



- 1 - Failed at interface of primer and tape adhesive
- 2 - Failed at interface of primer and iridite
- 3 - Failed at interface of tape adhesive and tape backing
- N - Approximate % of a given mode of failure.

THE

Calc.			REVISED	DATE	PEEL TEST DATA OF 3M'S Y-9050 TAPE	PAGE 2 of 2	DATA
Trac						TABLE I	
Chk							
Appr.					THE BOEING COMPANY SEATTLE 24, WASHINGTON		
Appr.							

EX-100-100000-2

[illegible]

DATA SHEET

NOTES:

- 1 - Failed at interface of primer and tape adhesive
- 2 - Failed at interface of primer and iridite
- 3 - Failed at interface of tape adhesive and tape backing
- N - Approximate % of a given mode of failure.