

SUBROUTINE DESCRIPTIONS AND LISTINGS FOR THE ORBIT DETERMINATION PROGRAM VOLUME I

Contract NAS5-9939

Prepared for:

GODDARD SPACE FLIGHT CENTER
GREENBELT, MARYLAND



PHILCO-FORD CORPORATION
Space & Re-entry Systems Division
Palo Alto, California

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Volume I

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SYSTEM DEVELOPMENT AND MISSION ANALYSIS DEPARTMENT
Space and Reentry Systems Division
Philco-Ford Corporation
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for

GODDARD SPACE FLIGHT CENTER
Greenbelt, Maryland

FOREWORD

Contract NAS5-9939 covered development of the Orbit Determination Program system described in this report, and several modifications to the Mark II Error Propagation Program developed under an earlier contract. Documentation of the modifications to the Mark II program was completed as additions to the Mark II manuals, and was delivered as each modification was completed. Final documentation of the Orbit Determination Program system is contained in

1. TR-DA1508 Program Description and Theoretical Basis for the Orbit Determination Program.
2. TR-DA1509 Subroutine Descriptions and Listings for the Orbit Determination Program.
3. TR-DA1510 Input - Output Summary for the Orbit Determination Program.

The program development was done in the Systems Development and Mission Analysis Department under the technical direction of R. E. Brown, Engineering Section Supervisor. Major technical contributors were:

W. S. Bjorkman	Senior Engineering Specialist
W. F. Colescott	Project Engineer
D. E. Ekman	Project Engineer
M. J. Brooks	Senior Programmer

ACKNOWLEDGEMENT

The contributions of Dr. Stanley F. Schmidt of Astro Consultants, Inc. are gratefully acknowledged. Dr. Schmidt is principally responsible for the filter design. He was also a major contributor in the definition of program capabilities and in the program conceptual design.

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SECTION 1

LABELLED COMMONS

1.0 INTRODUCTION

Labelled commons are used by the Orbit Determination Program system for all communication of data between subroutines where common usage is indicated. Blank common is used only for temporary storage within a given subroutine, and never for communication.

The data are assigned to the various commons according to the following rules:

1. Data required only by a set of functionally related subroutines are grouped into a functional common, referenced only by that set of subroutines.
2. Data required only by a given link are preferably stored in that link.
3. Data involving program dimensions which should be changeable are grouped according to the dimension involved.

This section contains the descriptions and block data listings of all labelled commons used by the ODP, arranged in alphabetical order. The descriptions contain general classifications of the data and detailed descriptions of the individual quantities. Where appropriate, references to paragraph numbers or equation numbers in the referenced reports are given in parenthesis.

/BUFCOM/BUFF1(40), KEY

No block data.

BUFCOM is used as a storage array from which to write the key record of each record pair on the output data tape. It also contains an output tape controlling switch KEY. KEY is not written on the tape.

/BUFCOM/BUFF1(40)

LOC	Symbol	Dimension	Definition
1	NPAIR		Record pair counter
2	NEOT		Beginning and end of tape indicator -1: First record after summary +1: Last record on tape 0: Other
3	NAMSTA		Name of receiving station
4	NMSTA 1		Name of transmitting station. In simulator NAMSTA = NMSTA1
5	NRCD		Arc record counter
6	NPTS		Number of time points in following record
7	KONT		Arc continuation key 1: Record pair is a continuation of previous record pair 0: Record pair initiates a new arc
8	MTYPE		Measurement system type indicator 1: C-Band 2: Goddard Range and Range Rate 3: Unified S-Band 4: DSIF (JPL)
9	NALIGN		Antenna alignment indicator for USBS 1: 30-foot dish, principal axis north 2: 85-foot dish, principal axis east
10	MODE		Doppler mode indicator 2: Two way 3: Three-way

BUFCOM-1

/BUFCOM/BUFF1 (40) (Continued)

LOC	Symbol	Dimension	Definition
11	DELT		Observation period, seconds
12	KTAU		Doppler n-count/t-count indicator 1: Count interval is fixed and found in TAU; data contains doppler counts 0: Doppler count is fixed and is found in TAU; data contains doppler count interval In simulator, KTAU = 1 always
13	ONTIME	d	Arc start time, seconds from 1950
15	TFIRST	d	Record start time, seconds from 1950
17	TLAST	d	Record stop time, seconds from 1950
19	TAU	d	Fixed doppler count interval or fixed count. See KTAU
21	FTR	d	Transmitter carrier frequency
23	C1	d	} Refraction constants
25	C2	d	
27	DR	d	Range ambiguity in units of range data
29	BIAS	d	Doppler offset bias
31	RETR	d	Transponder transmit/receive ratio
33	-----	d (4)	Not used by simulator

BUFCOM-2

/CROCOM/CCRO (86)

No block data.

CROCOM is used for communication between the polynomial fitting subroutine POLYFT and the simultaneous equation solver CROSIM. It contains the augmented matrix, a tag indicating the number of polynomial coefficients to be solved for, and a singularity flag. Subroutine CROSIM solves the equation $AX=C$. The augmented B matrix must be supplied as $\begin{bmatrix} A \\ C \end{bmatrix}$. The matrix A is destroyed in computation and the vector X is developed in place of C.

/CROCOM/CCRO (86)

LOC	Symbol	Dimension	Description
1	B	d(6, 7)	Augmented matrix. Matrix A must be stored in the first KK rows and columns of B. The vector C must be stored in the (KK+1)th column. The solution X is developed in column KK+1.
85	KK		Number of rows of B to use; number of elements of x to solve for.
86	ISING		Singularity flag, set by CROSIM. Set = 1 for singular A; set = 0 otherwise.

CROCOM-1

/DATCOM/CDAT (2400)

No block data.

DATCOM is used for communication between the four raw data decoding subroutines (CBTEST, GRTEST, SBTEST, DSTEST) and the polynomial fitting subroutine POLYFT of the TDEP. It contains two time tags, a quality indicator, and the values or four observables for up to 300 time points.

The size of the arrays in DATCOM arbitrarily causes any station pass containing more than 300 time points to be split into two or more station passes. If space permits, these arrays may be made larger by altering the DIMENSION and EQUIV-ALLENCE statements in subroutines CBTEST, GRTEST, SBTEST, DSTEST, and POLYFT, and also altering the value of the variable INPMAX set by a DATA statement in the subroutines CBTEST, GRTEST, SBTEST, and DSTEST.

/DATCOM/CDAT (2400)

Edited Data Array

LOC	Symbol	Dimension	Description
1	TIMTAG	d(300)	Data time tag, seconds from 1950.
601	TXTRA	(300)	Adjusted data time tag. $TXTRA(I) = TIMTAG(I) - TIMTAG(1)$
901	IQ	(300)	Quality indicator 0: Good data 1: Angles bad 2: Range bad 4: Doppler bad: > 4: Combination of above
1201	XANG	(300)	X-angle, azimuth, or hour-angle; radians.
1501	YANG	(300)	Y-angle, elevation, or declination; radians
1801	RANGE	(300)	Range; km or seconds
2101	DOP	(300)	Doppler; counts or seconds

DATCOM-1

/DATCOM/CDAT(299)

DATCOM is used for communication between the subroutines which compute the observables and the data processing routines of the DCP. It contains all the data which must be supplied for computation of the observables as well as the computed observables and their partial derivatives.

/DATCOM/CDAT(299)

Table 1

Measurement Data

Loc	Symbol	Dimension	Definition
1	BIAS	d(2)	Doppler bias frequency
			Spacecraft retransmission ratio
5	OBS	d(64)	See Table 2
133	FTR	d	Doppler transmission frequency
135	OMEGA	d	Earth rotation rate
137	SPDLT	d	Speed of light
139	STA	d(10)	See Table 3
159	TAU	d	Doppler count interval
161	TB2CO	d(18)	Transformations from B-frame to C-frame at signal reception by the receiving station, at end of doppler interval and at beginning of doppler interval
197	TB2CT	d(18)	Transformations from B-frame to C-frame at signal transmission by the transmitting station, at end of doppler interval and at beginning of doppler interval.
233	TT2BO	d(9)	Unit North, East, Down vectors at receiving station in the B-frame
251	TT2BT	d(9)	Unit vectors at the transmitting station
269	XV	d(12)	C-frame components of spacecraft position and velocity at end of doppler interval and at beginning of doppler interval

DATCOM-1

Table 1 - Continued

Loc	Symbol	Dimension	Definition
293	MLT		If = 1, speed of light partials are computed; if = 2, not computed.
294	MODE		Set = 2 or 3 for 2 or 3-way doppler, respectively
295	MSTA		If = 1, station location partials are computed; if = 2, not computed
296	MTIM		If = 1, time bias partials are computed; if = 2, not computed
297	NALIGN		X-Y mount alignment key: =1 for principal axis north, =2 for principal axis east.
298	NANG		If = 1, angles and their partials are computed; if = 2, not computed
299	NFRAC		If = 1, refraction effects are included; if = 2, not included

DATCOM-2

OBS d(64)

Table 2

Observables and Partial Derivatives

α_1 Azimuth X-Angle or Hour Angle	α_2 Elevation Y-Angle or Declination	ρ Ranging Observable	$\dot{\rho}$ Doppler Observable	Comments
1	17	33	49	
α_1	α_2	ρ	$\dot{\rho}$	Measurements
$\frac{\partial \alpha_1}{\partial x_v}$	$\frac{\partial \alpha_2}{\partial x_v}$	$\frac{\partial \rho}{\partial x_v}$	$\frac{\partial \dot{\rho}}{\partial x_v}$	Vehicle Position Partial (3)
$\frac{\partial \alpha_1}{\partial \dot{x}_v}$	$\frac{\partial \alpha_2}{\partial \dot{x}_v}$	$\frac{\partial \rho}{\partial \dot{x}_v}$	$\frac{\partial \dot{\rho}}{\partial \dot{x}_v}$	
				Vehicle Velocity Partial (3)
$\frac{\partial \alpha_1}{\partial \alpha_1}$	$\frac{\partial \alpha_2}{\partial \alpha_2}$	$\frac{\partial \rho}{\partial \dot{\rho}}$	$\frac{\partial \dot{\rho}}{\partial \omega_3}$	Measurement Bias Partial
$\frac{\partial \alpha_1}{\partial t}$	$\frac{\partial \alpha_2}{\partial t}$	$\frac{\partial \rho}{\partial t}$	$\frac{\partial \dot{\rho}}{\partial t}$	Observing Station Clock Bias Partial (total time derivative)
$\frac{\partial \alpha_1}{\partial c}$	$\frac{\partial \alpha_2}{\partial c}$	$\frac{\partial \rho}{\partial c}$	$\frac{\partial \dot{\rho}}{\partial c}$	Speed of Light Partial
$\frac{\partial \alpha_1}{\partial x_R}$	$\frac{\partial \alpha_2}{\partial x_R}$	$\frac{\partial \rho}{\partial x_R}$	$\frac{\partial \dot{\rho}}{\partial x_R}$	Receiving Station Position Partial (3)
0	0			Transmitting Station Position Partial (when operating in three-way doppler mode) (3)
0	0	$\frac{\partial \rho}{\partial x_T}$	$\frac{\partial \dot{\rho}}{\partial x_T}$	
0	0			

DATCOM-3

STA d(10)

TABLE 3
STATION DATA

Loc	Quantity
1-3	Receiving station position in the B-frame
4-5	Receiving station refraction coefficients.
6-8	Transmitting station position.
9-10	Transmitting station refraction coefficients.

DATCOM-4

/DCPCOM/CDCP(900)

Block data: Deck MC1378

DCPCOM is used for communication between the major links of the DCP. It contains those quantities which are required by two or more of the links, and those which are to be saved regardless of program overlay during a run.

/DCPCOM/CDCP(900)

Table 1

DCP Main Common

LOC	Symbol	Dimension	Definition
1	CON	d(8)	Universal constants (Table 2)
17	CBODY	(8, 11)	Central body constants (Table 3)
105	MAXZ		Maximum length of state vector (program dimension)
106	MAXQ		Maximum number of equation of motion parameters in the state
107	MAXD		Program dimension of equation of motion arrays (DQDCOM)
108	Spares	(2)	-----
110	YTEST		Value used for missing measurements on the data tape (-0.12345678E20)
111	IERR		Error identification key
112	ETAPE	(4)	Parameters defining the planetary ephemeris tape position (see DPFMRS)
116	NRSEND		Number of record pairs on the residual tape
117	NESEND		Number of record pairs on the estimation tape
118	NEST		Estimate tape record number used for constructing a <u>priori</u> estimate
119	TBEGIN	d	Time of first data on the data tape, seconds from 1950 Jan 0.0 ST
121	TEND	d	Time of last data on tape
123	STIMNX	(20)	Array of times of last data used in estimation, each station

DCPCOM-1

Table 1 - (Continued)

LOC	Symbol	Dimension	Definition
143	SN	d(13, 20)	Station names, locations, orientations with respect to the Earth-fixed reference frame
663	IFPLNT	(11)	Yes/No options for inclusion of planetary attractions in the equations of motions
676	IFOUTP	(11)	Output options (Table 4)
687	IFEMPS	(8)	Equation of motion options (Table 5)
695	NEMP	(8)	Variational equation dimensions (Table 6)
703	IFHARM	(24)	Yes/No options for inclusion of harmonic coefficient sensitivities in the variational equations
727	Spares	(4)	-----
731	NPROC	(22)	Process control options (Table 7)
753	Spares	(3)	-----
756	IPROC	(5)	Internal process indices (Table 8)
761	KEYOUT	(10)	Output control keys for trajectory integration (see OUTXPD)
771	NCB	(3, 2)	Harmonic summation limits for Earth, Moon, and body NBX
777	ICB		Current central body numbers
778	JCB		Central body type (1 = Earth, 2 = Moon, 3 = Other)
779	KCB		Next probable central body
780	NTBL		Number of integration points skipped in filling SBFCOM
781	STIMR	d(2)	Process start and stop times, seconds from 1950 January 0.0 ST
785	ETIMV1	d	Desired anchor point time, seconds from 1950 January 0.0 ET
787	NSCTRL	(14)	Data start control indices (Table 9)
801	DATEV1	d(3)	Date of desired anchor point (See DATINP)
807	RSTOP	d	Radius from body NTARGET for propagation stop
809	DRECT	d	Rectification test magnitude.

DCPCOM-2

Table 1 - (Continued)

LOC	Symbol	Dimension	Definition
811	EPSDC	d	Non-optional filter gain, ϵ
813	SCALE	d(6)	Scale factors for data start.
825	CONTROL	d(20)	Data start control parameters (Table 10).
865	Spares	(16)	-----
881	MSKIP	(4, 4)	Array of missing measurement keys.
897	RSTP2	d	(RSTOP) ²
899	CBRAN2	d	(Slant range from central body) ² at the anchor point.

DCPCOM-3

CON d(8)

Table 2

Universal Constants

Locations		Symbol	Math Symbol	Units	Block data value
CDCP	CON				
1	1	HPI	$\pi/2$		1. 5707963267948966
3	2	PI	π		3. 1415926535897932
5	3	TPI	2π		6. 2831853071795864
7	4	RTD	$180/\pi$	deg/rad	57. 295779513082329
9	5	DTR	$\pi/180$	rad/deg	0. 017453292519943295
11	6	SPMSD		sec/day	86400.
13	7	RSPMSD		day/sec	1. 1574074074074074x10 ⁻⁵
15	8	SPDLT	c	km/sec	299792. 5 (speed of light)

CBODY (8, 11)

Table 3

Central Body Constants

Loc	Units	Definition
1, i		i th body name
2, i	km	Radius of sphere of influence
3, i	km	Equatorial radius
4, i	km	Polar radius
5, i	rad/sec	Angular velocity
6, i	rad	Right ascension of the prime meridian, from the ascending node of the Earth's ecliptic on the body's equator
7, i	rad	Inclination of Earth's ecliptic with body's equator
8, i	rad	Longitude of the node, measured in the Earth's ecliptic from the Earth's mean vernal equinox of 1950. 0

DCPCOM-4

IFOUTP (11)

Table 4

Output Options

Location		Symbol	Affected Subroutines*	Definition
CDCP	IFOUTP			
676	1	NPROUT	OUTTRJ	Integration output key for the estimation/propagation link.
677	2	NRSOUT	OUTTR3	Integration output key for the residual link.
678	3	NCVOUT	COVOUT	Covariance matrix output key.
679	4	NESOUT	ESTMAT	Estimation process output key.
680	5	NSTTD	STTDAT	Output key for data start data selection.
681	6	NSTTF	STTFIT	Output key for data start polynomial smoothing.
682	7	NSTTG	STTBGN	Output key for data start initialization.
683	8	NSTTM	MXLEST	Output key for output during maximum likelihood estimation
684	9	NSTTR	STTINE	Output key for state output during data start.
685	10	NSTTS	STTEND	Output key for data start completion.
686	11	NDPØUT	ØUTTRJ ØUTTR3	Double precision output key.

* The values of IFOUTP(i) and their effects on output are described in the description of the named subroutines and in References 1 and 2.

IFEMP (8)

Table 5

Equation of Motion Options

Location		Symbol	Input Option*	Definition and values**
CDCP	IFEMP			
687	1	IFGRAV	X	Extra body harmonics included?
688	2	IFSOLR	X	Solar pressure included? 1 C_1 not in state 0 C_1 not in equations of motion -1 C_1 in state
689	3	IFATMD	X	Drag model included? 1 C_2, C_3 not in state 0 not in equations of motion -1 C_2 in state -2 C_3 in state -3 C_2, C_3 in state
690	4	IFVENT	X	Venting included? 1 C_4 not in state 0 not in equations of motion -1 C_4 in state
691	5	IFTIME		ET-UT parameters 0 γ_3, γ_4 not in state -1 γ_3 in state -2 γ_4 in state -3 γ_3, γ_4 in state
692	6	IFHARM		Harmonic key*** 0 no harmonics for current central body. ±4 zonal harmonics only ±5 tesseral harmonics
693	7	IFDRAG		Drag key*** 0 if central body is not NBD IFATMD if central body is NBD
694	8	IFBODY		IFATMD + IFHARM 0 only if acceleration is independent of central body orientation

* Yes/No options for inclusion in equations of motion (=1: yes).

** Values are computed by SETCAS

*** Values are computed by TRAJD, TRAJDP

DCPCOM-6

NEMP (8) *

Table 6

Variational Equation Dimensions

Location		Symbol	Description
CDCP	NEMP		
695	1	NDRAG	Location in RDEQ of drag parameters sensitivities
696	2	NHARM	Location in RDEQ of harmonic coefficient sensitivities
697	3	NSOLR	Location in RDEQ of solar pressure sensitivities
698	4	NVENT	Location in RDEQ of venting sensitivities
699	5	NTIME	Location in RDEQ of sensitivities to ET-UT
700	6	NPEND	First row of RDEQ not used for differential equations
701	7	NEMP3	3 x NEMPS
702	8	NEMP5	Number of equation of motion parameters

*The NEMP array is loaded by SETCAS during interpretation of the IFEMP array.

DCPCOM-7

NPROC (22)

Table 7

Process Control Options

Card Cols**	Field	NPROC LOC	Symbol*	Definition
1	I1	1	NPROCS	Process type specification
2	I1	2	NPROVR*	Does data follow the process card?
3-8	A6	3	NPRSTA	Name of receiving station
9-32	2(I3, I7, I2)	4	NDATE	Dates for process start and stop (dimensioned 3, 2)
33-36	I4	10	NPREST	Record number of <u>a priori</u> estimate
37-38	I2	11	NREWIND*	Rewind data tape before search?
			NPSTOP	Propagation stop event identification
39-40	I2	12	NPRSTO*	Store new estimate on tape?
			NTARGET	Target body number (propagation only)
41-42	I2	13	NPSKIP	Number of time points to be skipped between processed points
43-50	4I2	14	IFSKIP*	Is Ith measurement to be skipped? (dimensioned 4)
51-52	I2	18	IFBADD*	Is flagged "bad" data to be processed?
53-54	I2	19	IFOUTL*	Are "outliers" to be processed?
55-56	I2	20	NRTAPE*	Is residual tape to be written?
57-58	I2	21	NRPLOT*	Is residual plot desired?
59-60	I2	22	NRLIST	Residual output level option

*The options marked by an asterisk are yes/no options.

In each case, "no" is designated by a zero or blank field and "yes" by any non-zero integer.

** The card locations of the options are given here for completeness. For a more detailed description of the card and options, see Reference 1.

DCPCOM-8

IPROC (5)

Table 8

Internal Process Indices

Location		Symbol	Description
CDCP	IPROC		
756	1	IPROCS	NPROCS Mod 4 Process type, if any, to follow data start, if any.
757	2	ITRSTA	Location in NAMSTA of transmitting station.
758	3	INDSTA	Location in NAMSTA of receiving station.
759	4	IFTSTT	1 if process start time is specified, 2 if not.
760	5	IFTSTP	1 if process stop time is specified, 2 if not.

DCPCOM-9

NSCTRL (14)

Table 9

Data Start Control Indices

Location		Symbol	Description
CDCP	NSCTRL		
787	1	NPOINT	Number of points to be used for the maximum likelihood estimation.
788	2	NFPTS	Number of points to be used for polynomial fit.
789	3	NFSKIP	Number of points skipped between used points in the polynomial fit.
790	4	NFPOLY	Degree of polynomial +1.
791	5	NFERR	Number of points to shift for retry of polynomial fit.
792	6	MFTRY	Maximum number of trials allowed for polynomial fit.
793	7	MTRY	Maximum number of trials allowed for maximum likelihood estimation.
794	8	KSCALE	Scaling option < 0: scale factors are input in SCALE = 0: no scaling > 0: normalize XM
795	9	MAXPTS	Maximum number of points to be loaded into working array DATA.
796	10	MINPTS	Minimum number of points to be loaded.
797	11	MAXAMB	Maximum number of range ambiguity trials for m. l. e.
798	12	KAUTO	Logical yes/no (T/F) option for allowing re-trial with new data if m. l. e. is not completed with original data.
799	13	Spares	(2)

DCPCOM-10

CONTRL d(20)

Table 10

Data Start Control Parameters

Location		Symbol	Description
CDCP	CONTRL		
825	1	TFIT	Time from ONTIME to start polynomial fit.
827	2	EPSBAR	Convergence level for mean wgt sq of m. l. e. residuals.
829	3	EPSDEL	Fractional change in wgt sq of residuals.
831	4	EPSEUD	Test threshold for pseudo inverse.
833	5	FITLEV	Test factor for polynomial fit acceptance.
835	6	RCNST	Position constraint factor.
837	7	VCNST	Velocity constraint factor.
839	8	CNST	(1): Constraint vector control = 0: input G(I) in CNST (I+1) > 0: set G(I) on each trial < 0: set G(I) on first (2-7): Input constraint vector
853	15	CNTL	Convergence rate factor (β) control = 0: $\beta = \beta_0$ < 0: β variable downward only > 0: β variable
855	16	BO	β_0
857	17	BUP	β increase factor
859	18	BDOWN	β decrease factor
861	19	AUPPR	Maximum fractional correction for increasing β .
863	20	ALOWR	Minimum fractional correction.

```

$IBFTC MC1378 XR3,M94,NODD,LIST
BLOCK DATA
C DCP MAIN COMMON DCPC0001
C COMMON /DCPCOM/DCDP(900) DCPC0002
C DCPC0003
C DCPC0004
C DCPC0005
C DCPC0006
C DCPC0007
DOUBLE PRECISION CON (8)
1 ,CONTRL(20) ,RSTOP (2) DCPC0008
2 ,EPSDC ,SCALE (6) DCPC0009
C DCPC0010
DIMENSION CBODY(8,11) ,IFEMPS (8) ,KEYOUT(10) DCPC0011
1 ,NCB (3,2) ,IFOUTP(11) ,MSKIP(4,4) DCPC0012
2 ,IFPLNT(11) ,NSCTRL(14) DCPC0013
C DCPC0014
EQUIVFNCE (CDCP(663),IFPLNT) ,(CDCP(787),NSCTRL) DCPC0015
1 ,(CDCP( 17),CBODY ) ,(CDCP(761),KEYOUT) ,(CDCP(780),NTBL ) DCPC0016
2 ,(CDCP( 1),CON ) ,(CDCP(107),MAXD ) ,(CDCP(674),ITARGT) DCPC0017
3 ,(CDCP(825),CONTRL) ,(CDCP(106),MAXQ ) ,(CDCP(807),RSTOP ) DCPC0018
4 ,(CDCP(811),EPSDC ) ,(CDCP(105),MAXZ ) ,(CDCP(813),SCALE ) DCPC0019
5 ,(CDCP(687),IFEMPS) ,(CDCP(881),MSKIP ) ,(CDCP(110),YTEST ) DCPC0020
6 ,(CDCP(676),IFOUTP) ,(CDCP(771),NCB ) DCPC0021
C DCPC0022
C UNIVERSAL CONSTANTS DCPC0023
C CON(1)=PI/2 CON(2)=PI DCPC0024
C DATA CON / 1.5707963267948966 ,3.1415926535897932 DCPC0025
C CON(3)=2*PI CON(4)=RTD CON(5)=DTR DCPC0026
1 ,6.2831853071795864 ,57.295779513082329 ,.017453292519943295 DCPC0027
C CON(6)=SPMSD CON(7)=RSPMSD CON(8)=SPDLT DCPC0028
2 ,86400.00 ,1.1574074074074074D-5 ,299792.500 /DCPC0029
C DCPC0030
C CENTRAL BODY CONSTANTS DCPC0031
C BODY NAME DCPC0032
C DATA (CBODY(1,I),I=1,11)/ ,6HMFRCRY ,6HVENUS ,6HEARTH ,6HMARS DCPC0033
1 ,6HJUPTER ,6HSATURN ,6HURANUS ,6HNEPTUN DCPC0034
2 ,6HPLUTO ,6HSUN ,6HMOON /DCPC0035
C SPHERE OF INFLUENCE (KM) DCPC0036
C DATA (CBODY(2,I),I=1,11)/ 111000. ,616000. ,925000. ,565000. DCPC0037
1 ,480.E5 ,546.E5 ,520.E5 ,875.E5 DCPC0038
2 ,135.E5 ,1.E10 ,6600. /DCPC0039
C EQUATORIAL RADIUS (KM) DCPC0040
C DATA (CBODY(3,I),I=1,11)/ 2500. ,6200. ,6378.165 ,3400. DCPC0041
1 ,71350. ,60400. ,25550. ,25000. DCPC0042
2 ,2882. ,695550. ,1738. /DCPC0043
C POLAR RADIUS (KM) DCPC0044
C DATA (CBODY(4,I),I=1,11)/ 2500. ,6200. ,6356.5838,3400. DCPC0045
1 ,66600. ,54040. ,23725. ,24445. DCPC0046
2 ,2882. ,695500. ,1738. /DCPC0047
C ANGULAR VELOCITY (RAD/SEC) DCPC0048
C DATA (CBODY(5,I),I=1,11)/ 11*0./ DCPC0049
C RT ASC OF PRIME MERIDIAN (RAD) DCPC0050
C DATA (CBODY(6,I),I=1,11)/ 11*0./ DCPC0051
C OBLIQUITY (RAD) DCPC0052
C DATA (CBODY(7,I),I=1,11)/ 11*0./ DCPC0053
C LONGITUDE OF NODE (RAD) DCPC0054
C DATA (CBODY(8,I),I=1,11)/ 11*0./ DCPC0055
C DCPC0056
C PROGRAM DIMENSIONS AND CONSTANTS DCPC0057
C DATA MAXZ,MAXQ,MAXD/30,6,44/ DCPC0058
C DATA YTEST /- .12345678E20/ DCPC0059
C DCPC0060
C OPTIONS DCPC0061
C DCPC0062
C DATA IFPLNT/0, 1, 1, 1, 5*0, 1, 1/ DCPC0063
C DATA IFOUTP/2, 2, 1, 0, 6*10, 1/ DCPC0064
C DATA IFEMPS/0, 1, 1, 5*0/ DCPC0065
C DATA KEYOUT/0, 1, 3,-1, 6*0/ DCPC0066
C DATA NCB /4, 2, 0, 0, 2, 0/ DCPC0067
C DCPC0068
C PROGRAM CONTROL PARAMETERS DCPC0069
C DCPC0070
C DATA NSCTRL/ 30, 7, 1, 4, 4, 5, 10, 1,100, 0 DCPC0071
1 , 4, 0, 1, 0/ DCPC0072
C DATA RSTOP / 0.00, 1.0-1/ DCPC0073
C DATA EPSDC / 0.00/ DCPC0074

```

DATA SCALE / 6*1.DU/
DATA CONTRL/ 0.D+0, 1.D+1, 1.D-5, 1.D-8, 1.D+4
1 , 1.D-1, 1.D-1, -1.D+0, 6*0.D+0
2 , 1.D+0, 1.D+0, 2.D+0, 1.D-1, 1.3D+0, 1.D-8 /
DATA MSKIP /0,0,0,1, 0,0,0,0, 0,0,0,0, 0,0,1,0/
DATA ITARGT/4/
DATA NTRL /4/
END

DCPC0075
DCPC0076
DCPC0077
DCPC0078
DCPC0079
DCPC0080
DCPC0081

/DCRCOM/ZBAR(30), ZHAT(30), ZTOT(30), DCSAVE(74)

No block data.

DCRCOM is used for storing differential corrections and other quantities to be saved during linking in an estimation process. It is located in the DIFCOR link of the DCP. The differential corrections are:

- | | |
|------|--|
| ZBAR | The deviation of the "nominal" state at the anchor point from the <u>a priori</u> estimate. |
| ZHAT | The deviation of the current best estimate of state at the anchor point from the <u>a priori</u> estimate. |
| ZTOT | The deviation of the current <u>a priori</u> estimate from the original <u>a priori</u> estimate. |

The vectors ZBAR, ZHAT, ZTOT have dimensions MAXZ, and must be re-dimensioned when the maximum length of the state vector is changed. The affected subroutines are DIFCOR, ESTOUT and ESTMAT.

The array DCSAVE contains quantities set during initialization of ESTMAT to be saved during any linking. The quantities are described in the table below.

DCRCOM-1

DCSAVE (74)

ESTMAT Initialization Data

Loc	Symbol	Dimension	Description
1	OBIAS	d(4)	Measurement biases
9	YMOD	d(4)	Measurement moduli
17	EPSK	d(4)	Non-optimal filter gains (=EPSDC* measurement standard deviation)
25	Q2	d(4)	Measurement variances
33	DPREV	d	Square of position deviation from the nominal trajectory at last failure of the linearity test.
35	TFIRST	d	First station on-time, seconds from 1950 Jan 0.0
37	STPTIM		Process stop time, seconds from TFIRST
38	TREF		Start time for current data arc, seconds from TFIRST
39	LABEL	(12)	BCD measurement names, units
51	SYMBOL	(4)	Abbreviated measurement names
55	LSKIP	l (4)	Measurement skip options
59	LSTA	l	Internal switches
60	KOUTL	l	Internal switches
61	NDIM	(5)	Internally computed dimensions
66	NASGN	(9)	Permanent transfer addresses (for assigned "GO TO's")

DCRCOM-2

/DFMCOM/CDFM (2232)

Block data: Deck MC1324 or MC1344

DFMCOM is used for communication between DPFMRS and its using subroutines. The ephemeris tape record last used by DPFMRS is retained in DFMCOM to minimize tape searching.

DFMCOM (MC1324) and DPFMRS (MC1322) may be interchanged with DFMCOM (MC1344) and DPFMRS (MC1342) for space savings, with some reduction of ephemeris capabilities (see DPFMRS). Dimensions for the arrays in DFMCOM for both deck pairs are given in the table below. Other decks referencing DFMCOM need not be compiled provided DFMCOM is loaded before those decks.

/DFMCOM/CDFM(2232)
*CDFM(1524)

Planetary Ephemeris Storage

Loc	Symbol	Dimension	Description
1	IFM	(14)	Ephemeris output options 0: no output for the i th body 1: output position only 2: output position and velocity
15	RFM	d(6, 12)	Planetary positions and velocities
159	DFM	d(4)	Nutations in longitude and obliquity and nutation rates
167	BFM	d(829) d(577)	Position-velocity buffer
1825	SNT	(2, 204) (2, 102)	Nutation buffer

*Where two dimensions are given, the first refers to decks MC1322, MC1324, and the second refers to MC1342, MC1344.

DFMCOM-1

```
$IBFTC MC1344 XR3,M94,NODD,LIST
BLOCK DATA
DOUBLE PRECISION   BFM,DFM,RFM
COMMON             /DFMCOM/IFM(14),RFM(6,12)
1                  ,DFM (4),BFM (577),SNT(2,102)
DATA IFM/0,0,2,7*0,2,2,1,0/
DATA SNT/204*0./
END
```

FMBD0001
FMBD0002
FMBD0003
FMBD0004
FMBD0005
FMBD0006

```
$IBFTC MC1324 XR3,M94,NODD,LIST
BLOCK DATA
DOUBLE PRECISION   BFM,DFM,RFM
COMMON             /DFMCOM/IFM(14),RFM(6,12)
1                  ,DFM (4),BFM (829),SNT(2,204)
DATA IFM/0,0,2,7*0,2,2,1,0/
DATA SNT/408*0./
END
```

FMBD0001
FMBD0002
FMBD0003
FMBD0004
FMBD0005
FMBD0006

/DQDCOM/CDQD (26N + 28)

No block data.

DQDCOM is used for communication between DEQD and the remainder of the propagation link of the DCP. Its size depends on the maximum number of variational equations to be allowed. If

M_Q = Maximum number of equation of motion parameters permitted in the state vector, then

$N = 3M_Q + 26$

For example, if up to six parameters are to be permitted simultaneously, $N = 44$, and DQDCOM has length 1172. The values of M_Q , N used must be compiled into MAXQ, MAXD of DCPCOM. Changes in these values require recompilation of

Subroutine	Deck
DEQD	MC133D
TRAJDP	MC133F
ACCTRJ	MC133A
OUTTRJ	MC1327

See also SBFCOM.

DQDCOM-1

Table 1
Principal Arrays

Location	Name	Dimension	Definition
1	NDEQ	10	Array of fixed-point integration control parameters.
11	CDEQ	10	Array of floating-point control parameters.
21	XDEQ	d(4)	Independent variable values.
29	ADEQ	d(N)	Second derivatives and stopping functions.
2N+29	RDEQ	d(N)	Dependent variables.
4N+29	VDEQ	d(N)	First derivatives
6N+29	FDEQ	d(N, 10)	Storage array used by DEQD .

CDEQ

Table 2

Floating-Point Parameters

Component	Symbol	Definition
1	H	Current integration interval.
2	HP	Previous integration interval.
3	HO	Starting interval size.
4	HU	Upper limit for doubling interval size.
5	HL	Lower limit for halving interval size.
6	HE	Interval for even-interval output.
7	EU	Truncation error upper limit.
8	EL	Truncation error lower limit.
9-10		Not used.

DQDCOM-2

NDEQ
Table 3
Fixed-Point Parameters

Component	Symbol	Definition and Values
1	NI	Integration mode option 0: Kutta Mode 1: Adams Mode
2	NO	Output option 0: Output special points only. ≠ 0: Output every NOth integration point.
3	NS	Number of stopping functions (≤ 5).
4	NA	Interval control option 0: fixed interval size ≠ 0: interval size controlled by accuracy testing
5	NC	Predictor-corrector option (Adams integration) 0: open formulas 1: predictor-corrector without recomputation of corrected second derivatives. 2: predictor-corrector
6	NE	Number of equations to be integrated.
7	NT	Number of equations + number of stopping functions.
8	IP	Point identification key 0: regular integration point 1: even-interval output point 3: stop point
9	IT	Row number of stopping function causing stop.

Note: Components 1-6 are supplied prior to the call of DEQD, and components 7-9 are supplied by DEQD.

DQDCOM-3

XDEQ (4)

Table 4

Independent Variable

Component	Symbol	Definition
1	X	Running value of the independent variable.
2	XF	Final value of the independent variable.
3	XE	Next even-interval output point.
4	XO	Value of X at start of current integration interval.

DQDCOM-4

/EDTCOM/INDDAT(40), BUFDAT(85, 6)

No block data.

EDTCOM is used for the storage of the identification (INDDAT) and data (BUFDAT) records from the edited data tape in the DCP. The contents of these records are described below.

INDDAT(40)
Data Identification Array

Location	Symbol	Dimension	Description
1	NEDT		Record pair count
2	NEOT		End-of-tape indicator -1 if NEOT = 1 +1 if tape-end record 0 otherwise
3	NRSTA		Name of observing station
4	NTSTA		Name of transmitting station
5	NRCD		Arc record count, 1 on first pair for each data arc
6	NPTS		Number of time points in the block
7	KONT		Continuation key 0 if this block begins the arc 1 if another block precedes
8	MTYPE		Measurement type
9	NALIGN		X-Y mount orientation key
10	MODE		Doppler mode, 2- or 3-way
11	DELT		Measurement Time interval

EDTCOM-1

INDDAT(40) (continued)

Location	Symbol	Dimension	Description
12	KTAU		Count mode 1 fixed interval (in TAU) 0 fixed count (in TAU)
13	ONTIME	d	Arc start time, seconds from 1950
15	TFIRST	d	Block start time
17	TLAST	d	Block end time
19	TAU	d	Fixed count or interval
21	FTR	d	Transmitter frequency
23	CREFR	d(2)	Refraction coefficients
27	DRNGE	d	Range ambiguity
29	OMG3	d	Bias frequency
31	OMG4	d	Retransmission ratio
33	NBAD	(4)	Number of bad points, each measurement
37	SSDEV	(4)	Sample standard deviations

The BUFDAT array is a single precision array of dimension (85, 6) containing the edited data as read from the data tape. Each row contains the following data in the six columns:

BUFDAT (I, 1) = Quality indicator
 (I, 2) = Time, seconds from ONTIME
 (I, 3) = Angle 1
 (I, 4) = Angle 2
 (I, 5) = Range
 (I, 6) = Doppler

EDTCOM-2

/DQ3COM/CDQ3(236)

No block data.

DQ3COM is used for communication between DEQ3 and the remainder of the residual link of the DCP. It is referenced by

Subroutine	Deck
DEQ3	MC133E
DTRAJ	MC133G
TR3ACC	MC133B
TR3OUT	MC1328

Principal Arrays

Location	Name	Dimension	Definition
1	NDEQ	10	Array of fixed-point integration control parameters.
11	CDEQ	10	Array of floating-point control parameters.
21	XDEQ	d(4)	Independent variable values.
29	ADEQ	d(8)	Second derivatives and stopping functions.
45	RDEQ	d(8)	Dependent variables.
61	VDEQ	d(8)	First derivatives.
77	FDEQ	d(8, 10)	Storage array used by DEQ3.

The components of the arrays are the same as those described in DQDCOM. DQ3COM is limited to three differential equations, however, and five stopping functions.

DQ3COM-1



Missing measurements at a given time are filled with the constant $-0.12345678E20$. Measurements which are rejected by the TDEP as outliers are flagged by a minus sign.

The quality indicator is the sum of three indicators, $I_1 + I_2 + I_4$, where each indicator is zero if the corresponding measurement values are valid, and non-zero if the measurements are missing or flagged "bad" by the tracking station operator or are garbled in transmission. The indicators are;

$I_1 = 1$ if angle data is bad or missing,

$I_2 = 2$ if range data is bad or missing,

$I_4 = 4$ if doppler data is bad or missing.

EDTCOM-3

/EQMCOM/CEQM(2214)

Block data: Deck MC1376

EQMCOM is used for the storage of equation of motion parameters, their standard deviations and correlation coefficients. It is used by SETCAS in constructing the equation of motion portions of the a priori estimate and covariance matrix.

EQMCOM may be modified by overlay input at the case level (see References 1 and 2).

/EQMCOM/CEQM(2214)

Location	Symbol	Dimension	Description
1	SPCDAT	d(31)	Spacecraft position and velocity, values, standard deviations and normalized correlations.
63	EFEDAT	d(83)	Planetary gravitational constants, a. u., R_E .
229	PREDAT	d(15)	Solar pressure, drag (2), venting.
259	DELDAT	d(5)	ET-UT parameters (2).
269	EHADAT	d(324)	Earth harmonic coefficients.
917	MHADAT	d(324)	Moon harmonic coefficients.
1565	XHADAT	d(325)	Extra body harmonic coefficients.

EQMCOM-1

```

SIBFTC MC1376 XR3,M94,NODD,LIST
BLOCK DATA
C EQUATION OF MOTION BLOCK DATA EQMC0001
COMMON /EQMCOM/EQMC(2214) EQMC0002
DOUBLE PRECISION SPCDAT(31), EFEDAT(83), PREDAT(15) EQMC0003
1, DELDAT(5), EHADAT(324), MHADAT(324), XHADAT(325) EQMC0004
EQUIVALENCE (EQMC(1),SPCDAT), (EQMC(63),EFEDAT) EQMC0005
1, (EQMC(229),PREDAT), (EQMC(259),DELDAT), (EQMC(269),EHADAT) EQMC0006
2, (EQMC(917),MHADAT), (EQMC(1565),XHADAT) EQMC0007
C EQMC0008
C SPACECRAFT POSITION (KM) EQMC0009
DATA (SPCDAT(I),I=1,31)/.20947126D4,-.62193760D4,-.32542986D1 EQMC0010
C EQMC0011
C SPACE CRAFT VELOCITY (KM/SEC) EQMC0012
1,.65028256D1,.21881185D1,.36959843D1 EQMC0013
C EQMC0014
C POSITION AND VELOCITY STANDARD DEVIATIONS (KM, KM/SEC) EQMC0015
2,3*1.D-2,3*.333333333333333D-2 EQMC0016
C EQMC0017
C POSITION AND VELOCITY CORRELATIONS EQMC0018
3,15*0.D0 EQMC0019
C EQMC0020
C CENTRAL BODY AND TIME OF ESTIMATE (YM. DHM. SEC.) EQMC0021
4, 3.D0, 6701.D0, 10000.D0, 0.D0/ EQMC0022
C EQMC0023
C PLANETARY MU-S (KM**3/SEC**2) EQMC0024
MERCURY VENUS EARTH EQMC0025
DATA (EFEDAT(I),I=1, 83)/21685.53D0, 324853.4D0, 398603.2D0 EQMC0026
C EQMC0027
C MARS JUPITER SATURN URANUS NEPTUNE EQMC0028
1,42977.8D0, .1267106D9, .379187D8, .5803292D7, .7026072D7 EQMC0029
C EQMC0030
C PLUTO SUN MOON BARYCENTER EQMC0031
2,.3317886D6, .132715445D12, 4902.779D0, .40350579D6 EQMC0032
C EQMC0033
C AU(KM) AND MEAN EARTH RADIUS(KM) EQMC0034
3,.149599D9, 6378.3255D0 EQMC0035
C EQMC0036
C PLANETARY MU STANDARD DEVIATIONS (KM**3/SEC**2) EQMC0037
MERC VENUS EARTH MARS JUPITER SATURN URANUS EQMC0038
4,130.D0, 160.D0, 150.D0, 150.D0, 381.D1, 228.D1, 87.D3 EQMC0039
C EQMC0040
C NEPTUNE PLUTO SUN MOON EQMC0041
5,1.05D5, 2475.D1, 4.D7, 25.D0 EQMC0042
C EQMC0043
C THREE SPARE LOCATIONS SAVED HERE EQMC0044
6,3*0.D0 EQMC0045
C EQMC0046
C CORRELATIONS EQMC0047
7, 55*0.D0/ EQMC0048
C EQMC0049
C VALUES OF SOLAR PRESS DRAG1 DRAG2 VENT EQMC0050
DATA (PREDAT(I),I=1,15)/.94D7, .109D0, .14294D0, 0.D0 EQMC0051
C EQMC0052
C STANDARD DEVIATIONS EQMC0053
1, .94D6, .109D-1, .1429D-1, 1.D-6 EQMC0054
C EQMC0055
C CORRELATIONS CENTRAL BODY FOR DRAG EQMC0056
2, 6*0.D0, 3.D0/ EQMC0057
C EQMC0058
C VALUES OF ET-UT RATE, STANDARD DEV-S CORRS EQMC0059
DATA (DELDAT(I),I=1,5)/35.D0, 0.D0, 2.D0, 1.D-5, 0.D0/ EQMC0060
C EQMC0061
C EARTH HARMONICS J20 J30 J40 J50 EQMC0062
DATA (EHADAT(I),I=1,324)/.1082D-2, -.23D-5, -.18D-5, 0.D0 EQMC0063
C EQMC0064
C J60 J70 J21 L21 J22 L22 J31 EQMC0065
1,0.D0, 0.D0, 0.D0, 0.D0, -.12D-5, -.46D0, -.19D-5 EQMC0066
C EQMC0067
C L31 J32 L32 J33 L33 J41-L44 EQMC0068
2,.08D0, -.14D-6, -.293D0, -.1D-6, .743D0, 8*0.D0 EQMC0069
C EQMC0070
C STANDARD DEVIATIONS J20 J30 J40 J50-J70 EQMC0071
3, .108D-3, .23D-6, .18D-6, 3*.1D-6 EQMC0072
C EQMC0073
C EQMC0074

```

C	J21	L21	J22	L22	J31	L31	J32	EQMC0075
	4,.1D-6,	.5D-1,	.12D-6,	.5D-1,	.19D-6,	.5D-1,	.14D-7	EQMC0076
C								EQMC0077
C	L32	J33	L33	J41	L41	J42	L42	EQMC0078
	5,.5D-1,	.1D-7,	.5D-1,	.1D-7,	.5D-1,	.1D-7,	.5D-1	EQMC0079
C								EQMC0080
C	J43	L43	J44	L44				EQMC0081
	6,.1D-7,	.5D-1,	.1D-7,	.5D-1				EQMC0082
C								EQMC0083
C	CORRELATIONS							EQMC0084
	7,276*0.D0/							EQMC0085
C	MOON HARMONICS							EQMC0086
			J20		J30-J70			EQMC0087
			DATA (MHADAT(I),I=1,324)/.207107D-3,		5*0.D0			EQMC0088
C								EQMC0089
C	J21	L21	J22	L22	J31-L44			EQMC0090
	1,0.D0,	0.D0,	-.20716D-4,	0.D0,	14*0.D0			EQMC0091
C								EQMC0092
C	STANDARD DEVIATIONS							EQMC0093
			J20		J30-J70			EQMC0094
			2,	.207107D-4,	5*1.D-6			EQMC0095
C								EQMC0096
C	J21	L21	J22	L22	J31	L31	J32	EQMC0097
	3,1.D-6,	.5D-1,	.20716D-5,	.5D-1,	.1D-6,	.5D-1,	.1D-6	EQMC0098
C								EQMC0099
C	L32	J33	L33	J41	L41	J42	L42	EQMC0100
	4,.5D-1,	.1D-6,	.5D-1,	.1D-6,	.5D-1,	.1D-6,	.5D-1	EQMC0101
C								EQMC0102
C	J43	L43	J44	L44				EQMC0103
	5,.1D-6,	.5D-1,	.1D-6,	.5D-1				EQMC0104
C								EQMC0105
C	CORRELATIONS							EQMC0106
	6,276*0.D0/							EQMC0107
C	EXTRA BODY HARMONICS							EQMC0108
			J20-J70		J21-L44			EQMC0109
			DATA (XHADAT(I),I=1,325)/6*0.D0,		18*0.D0			EQMC0110
C								EQMC0111
C	STANDARD DEVIATIONS							EQMC0112
			J20-J70	J21	L21	J22	L22	EQMC0113
			1,	6*.1D-6,	.1D-6,	.5D-1,	.1D-6,	.5D-1
C								EQMC0114
C	J31	L31	J32	L32	J33	L33	J41	EQMC0115
	2,.1D-6,	.5D-1,	.1D-6,	.5D-1,	.1D-6,	.5D-1,	.1D-6	EQMC0116
C								EQMC0117
C	L41	J42	L42	J43	L43	J44	L44	EQMC0118
	3,.5D-1,	.1D-6,	.5D-1,	.1D-6,	.5D-1,	.1D-6,	.5D-1	EQMC0119
C								EQMC0120
C	CORRELATIONS							EQMC0121
	4,276*0.D0							EQMC0122
C	CENTRAL BODY NUMBER							EQMC0123
	5,4.D0/							EQMC0124
	END							

/ESOCOM/CESO(804), DESO($2N^2+4N+294$)

No block data.

ESOCOM is used only by SETCAS for the temporary storage of the estimate requested from tape by the case card. The arrays CESO and DESO are images of the arrays CEST (ESTCOM) and DEST (ES1COM). DESO and its component arrays must be redimensioned when ES1COM is redimensioned to preserve the correspondence between DESO and DEST.

ESOCOM-1

/ESRCOM/CESR (304)

No block data.

ESRCOM is used by the residual link of the DCP to permit overlay of state elements before computation of residuals. It contains an image of the first 244 elements of ESTCOM, as well as SER(14, 2) containing the columns of SE(ESTCOM) corresponding to the receiving station (INDSTA) and transmitting station (ITRSTA), and CRF(2) containing refraction coefficients to be used in lieu of those of the edited data tape.

ESRCOM-1

/ESTCOM/CEST(804)

No block data.

ESTCOM is used for storing the current estimate of the quantities which may be included in the extended state vector and certain quantities required for their interpretation. It is loaded in the main (0) link of the DCP (see also EQMCOM, SETCAS).

/ESTCOM/CEST(804)

Table 1

Location	Symbol	Dimension	Definition
1	NESPOS		Estimate number; estimation tape record count.
2	NBY		Central body number at anchor point.
3	NBD		Body with drag parameters in PREDAN.
4	NBX		Body with harmonics stored in XHADAN.
5	NBH		Body with harmonics included in state.
6	NUMSTA		Number of stations on tape.
7	NAMSTA	(20)	Names of stations on tape.
27	KOUNTN		Number of state elements.
28	Spare		-----
29	ETIMVA	d	Vehicle ephemeris time at anchor point, seconds from 1950 January 0.0 ET.
31	TX2Z	d(3, 3)	Transformation, C-frame to coordinates of state.
49	SPCDAN	d(6)	Vehicle position and velocity relative to Earth at anchor point, C-frame.

ESTCOM-1

/ES1COM/DEST($2N^2+4N+294$)

No block data.

ES1COM is used for the storage of the covariance matrix and certain associated indices and treatment codes. It is loaded in the SETCAS link, and is dumped on the scratch tape, unit 09. It is read into ES2COM in the DIFCOR link of the DCP.

The dimensions of the arrays in ES1COM depend on N, the maximum permissible length of the state vector. The dimensions of the arrays and their locations in DEST are given as functions of N in the table below. Changes in N require recompilation of the following subroutines which reference the commons. The value used for N must be compiled into MAXZ in DCPCOM (See also DCRCOM).

	Subroutine	Deckname
ES1COM	SETCAS	MC13SW
	BLDCOV	MC13SX
	COVOUT	MC13CV
ES2COM	ESTMAT	MC133J
	ESTOUT	MC1330
	UPDATP	MC138A
	COVOUP	MC138B
ES3COM	STTEND	MC13S5

ES1COM-1

/ES1COM/DEST($2N^2 + 4N + 294$)

General		N = 30		Array	Definition
Location	Dimension	Loc	Dimension		
1	d(N, N)	1	d(30, 30)	PNEW	Covariance matrix
$2N^2 + 1$	(N)	1801	(30)	STNAMN	Output names of state elements
$2N^2 + N + 1$	(N)	1831	(30)	TRAKER	Output names of stations for each state element
$2N^2 + 2N + 1$	(N)	1861	(30)	ITRETN	State element treatment codes 1: element to be estimated 0: element to be included only
$2N^2 + 3N + 1$	(N)	1891	(30)	ILOCN	State element identification keys
$2N^2 + 4N + 1$	(54)	1921	(54)	KLOCN	State group identification keys
$2N^2 + 4N + 55$	(12, 20)	1975	(12, 20)	NSN	Station error treatment codes

ES1COM-2

/EXIC/W(36), CRX(9, 2)

No block data.

EXIC is used as an input buffer for the reading of the trajectory tape by the TDS. The array W contains interpolation coefficients for the computation of vehicle state at a given time (see EXINST). The array CRX contains the "critical event" record from the trajectory tape, defining data relevant to initialization, patching, etc.

EXIC-1

/INPCOM/C(700)

Block Data: Deck MC13IP

INPCOM is used for storage and communication of physical and mathematical constants by the TDS. It is also used in part as additional working common. The values compiled in the block data subroutine may be altered by use of the overlay feature of the program. Normally, values beyond C(117) should not be overlaid, since they are used as working common and the required values are generated by the program.

/INPCOM/C(700)

Table 1

LOC	Symbol	Dimension	Description
1	HPI		One-half pi
2	PI		Pi
3	TWPI		Two pi
4	RTD		Radian to degree conversion factor
5	DTR		Degree to radian conversion factor
6	SPMSD		Seconds per mean solar day
7	RSPMSD		Reciprocal of seconds per mean solar day
8	AU		Kilometers per astronomical unit
9	SPDLT		Speed of light in km/sec
10	STPSZ		Step size constant used by SBEV2
11	BODC	(10, 8)	Planetary constants. See Table 2
91	Spares	(21)	-----
112	XMIN		Minimum acceptable length of data arcs (station passes) in seconds. Any arc shorter than XMIN will be eliminated

INPCOM-1

Table 1 (Continued)

LOC	Symbol	Dimension	Description
113	XMAX		Maximum acceptable length of data arcs in seconds. Any arc longer than XMAX will be divided into two or more arcs.
114	RDELAY		When the spacecraft appears above the horizon, the first measurement will be delayed by a random interval computed as the absolute value of a uniform distribution with mean zero, standard deviation of RDELAY. May be set to zero if desired.
115	XTROUT		Extra output key. If 0., only summary data will be written on the system output tape. If 1., all measurement values will be written out.
116	CFRAC1		} Refraction constants
117	CFRAC2		
118	Spares	(82)	-----
200	S	(23, 12)	<p>Array of tracking station parameters. Each column corresponds to a station number. Rows are assigned as follows:</p> <ol style="list-style-type: none"> 1 Station name, BCD 2 Observation interval, seconds 3 Geodetic latitude, radians 4 Longitude, radians 5 Altitude, kilometers 6 Artificial horizon, radians 7 Artificial zenith, radians 8 Random range error standard deviation, kilometers 9 Random doppler error std. dev., counts/sec 10 Random angle 1 error std. dev., radians 11 Random angle 2 error std. dev., radians 12 Range bias, km 13 Doppler bias, counts/sec 14 Angle 1 bias, radians 15 Angle 2 bias, radians 16 Northing error, km 17 Easting error, km 18 Down error, km 19 Clock bias, seconds

INPCOM-2

Table 1 (Continued)

LOC	Symbol	Dimension	Description
(200 Cont.)			
476	B	(91)	<p>20 Doppler offset bias, Hz</p> <p>21 Transponder retransmission ratio, (transmit/receive)</p> <p>22 Transmitter carrier frequency, Hz</p> <p>23 Doppler count interval, seconds</p> <p>Array of beacon parameters. The first 80 elements may be considered as an 8 x 10 array where each column corresponds to a beacon number. Rows are as follows:</p> <p>1 Name, BCD</p> <p>2 Latitude, radians</p> <p>3 Longitude, radians</p> <p>4 Altitude, kilometers</p> <p>5 Artificial horizon, radians</p> <p>6 Latitude error, km</p> <p>7 Longitude error, km</p> <p>8 Altitude error, km</p> <p>The remaining 11 locations are:</p> <p>81 Observation period, seconds</p> <p>82 Spare</p> <p>83 Random range error std. dev., km</p> <p>84 Random range rate error, std. dev., km/sec</p> <p>85 Random angle 1 error std. dev., radians</p> <p>86 Random angle 2 error std. dev., radians</p> <p>87 Range bias, km</p> <p>88 Range rate bias, km/sec</p> <p>89 Angle 1 bias, radians</p> <p>90 Angle 2 bias, radians</p> <p>91 Time or clock bias, seconds</p>
556	DELBEC		DELBEC = B(81)
567	OB	(18)	<p>Onboard measurement data as follows:</p> <p>1 Random height error std. dev., km</p> <p>2 Random height rate error std. dev., km/sec</p> <p>3 Height bias, km</p> <p>4 Height rate bias, km</p> <p>5 Time bias, seconds</p>

INPCOM-3

Table 1 (Continued)

LOC	Symbol	Dimension	Definition
			6 Altitude to cease radar observations, km
			7 Period of radar measurements
			8 Angle 1 error k, where:
			$\sigma_e^2 = k_1^2 + k_2^2 + k_2^2 \left(2 \sin^{-1} \frac{r_p}{r}\right)^2$
			k_1 in radians
			9 Angle 1 k_2 , dimensionless
			10 Angle 2 k_1
			11 Angle 2 k_2
			12 Subtended angle k_1
			13 Subtended angle k_2
			14 Period of angular measurements, seconds.
			15 Right ascension of reference star, radians.
			16 Declination of reference star, radians.
	Spare		17 -----
	Spare		18 -----
573	DELRAD		DELRAD = OB(7)
580	DELOPT		DELOPT = OB(14)
585	EMP	(24)	Equation of motion parameters, not now used
609	Spares	(64)	-----
673	IKAS		Case counter
674	Spares		-----
675	HEAD	(24)	HEAD(1-12) is a 12 word alphanumeric header used to identify the trajectory tape on tape 10. HEAD (13-23) is an 11-word alphanumeric header written on the output data tape HEAD (24) is not used

INPCOM-4

Table 1 (Continued)

LOC	Symbol	Dimension	Definition
699	ITAPE		Key for data tape. If ITAPE \neq 0, then no output tape will be written
700	IOCAS		Not used

BOCD (8, 10)

Table 2

The first 7 columns correspond to celestial bodies as follows:

1	Earth
2	Moon
3	Sun
4	Venus
5	Mars
6	Saturn
7	Jupiter

The remaining 3 columns are reserved for extra bodies. The row assignments are as follows:

1	Name, BCD
2	Gravity constants, km^3/sec^2
3	Semi-major axis, km
4	Semi-minor axis, km
5	Sphere of influence, km
6	Rotation rate, rad/sec
7	STEDT step size, seconds, not used
8	Interpolation interval, days, not used

INPCOM-5

\$IBFTC MC13IP NOREF,M94,NODD,XR3
CMC13IP INPCOM BLOCK DATA FOR MESERP

```
      BLOCK DATA
      COMMON/INPCOM/C(700)
      DIMENSION BODC(10,8), CON(10)
      3, S(23,12), B(91), OB(18), EMP(24), HEAD(24)
      EQUIVALENCE (C(1),CON), (C(11),BODC)
      1, (C(112),XMIN), (C(113),XMAX), (C(114),RDELAY)
      4, (C(675),HEAD)
      3, (C(200),S), (C(476),B), (C(567),OB), (C(585),EMP)
C (C(673),IKAS)
C (C(699),ITAPE)
C (C(700),IOCAS)
      C C(1)= PI/2 C(2)= PI C(3)= 2*PI
      DATA(CON(I),I=1,10)/1.57079633,3.14159265,6.2831853
      C C(4)= RTD C(5)= DTR C(6)= SPMSD C(7)= RSPMSD C(8)= KM/AU
      1,57.2957796,.0174532926,86400.,1.15740741E-5,.149599E9
      C C(9)= SPD OF LIGHT C(10)= SBEV CONSTANT FOR STEPSIZE FORMULA
      2,299774.,.16/
      C NAMES OF BODIES
      DATA (BODC(I,1),I=1,10)/5HEARTH,4HMOON,3HSUN,5HVENUS,4HMARS,
      16HSATURN,6HJUPITR,6HBODY 8,6HBODY 9,6HBOD 10/
      C GRAVITY CONSTANTS
      DATA(BODC(I,2),I=1,7)/398603.2,4900.7588.,.13271545E12,324769.5,
      142977.8,3.791870E7,1.267106E8/
      C SEMI-MAJOR AXIS
      DATA(BODC(I,3),I=1,7)/6378.165,1738.,695500.,6200.,3400.,60400.
      1,71350./
      C SEMI-MINOR AXIS
      DATA(BODC(I,4),I=1,7)/6356.5838,1738.,695500.,6200.,3400.,54050.
      1,66600./
      C SPHERE OF INFLUENCE (KM)
      DATA (BODC(I,5),I=1,7)/925000.,66000.,1.E10,616000.,565000.
      1.,546E8.,.48E8/
      C ROTATION RATES (RAD/SEC)
      DATA (BODC(I,7),I=1,7)/.72921152E-4, .266169952E-5, 0.0
      1, 0.0 ,.70882177E-4 ,.170553347E-3 ,.177491110E-3/
      C MAX STEP SIZE (SEC)
      DATA (BODC(I,6),I=1,7)/43200. ,21600. ,864000. ,43200.
      1,43200. ,86400. ,172800./
      C INTERPOLATION INTERVALS
      DATA (BODC(I,8),I=1,7)/7*40./
      DATAXMIN,XMAX,RDELAY/ 5., 43200., 10. /
      DATA (C(I),I=91,111)/21*0./
      C EXTRA OUTPUT KEY
      C REFRACTION CONSTANTS
      DATA (C(I),I=115,117)/0.,.34E-3,.138771E-3/
      DATA (C(I),I=118,700)/583*0./
      END
```

BLOK0001
BLOK0002
BLOK0003
BLOK0004
BLOK0005
BLOK0006
BLOK0007
BLOK0008
BLOK0009
BLOK0010
BLOK0011
BLOK0012
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BLOK0014
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BLOK0032
BLOK0033
BLOK0034
BLOK0035
BLOK0036
BLOK0037
BLOK0038
BLOK0039
BLOK0040
BLOK0041
BLOK0042
BLOK0043
BLOK0044
BLOK0045
BLOK0046
BLOK0047
BLOK0048

/MESCOM/CMES(510)

No block data.

MESCOM is used for the transmission of output edited data to tape. It is written as the second record of each record pair on the edited data tapes. Subroutine POLYFT lifts data from DATCOM and transfers it to MESCOM for output in records of 85 time points. The units of the observables are unchanged from those of DATCOM. MESCOM is also used by subroutines MIXIT, EGRESS, and SCANIT.

/MESCOM/CMES(510)

Data Output Buffer

LOC	Symbol	Dimension	Description
1	XMES	(85, 6)	XMES (I, 1) Quality indicator 0: good data 1: Angles bad 2: Range bad 4: Doppler bad > 4: Combination of above XMES (I, 2) Time tag, seconds from start of arc XMES (I, 3) X, azimuth, or hour angle, radians XMES (I, 4) Y, elevation, or declination, radians XMES (I, 5) Range, km or seconds XMES (I, 6) Doppler, km or seconds

MESCOM-1

/MIXCOM/IMIX(4)

No block data.

MIXCOM is used for communication of tape handling information between the MAIN program and the merging subroutine MIXIT and the output subroutine EGRESS. It contains the tape unit numbers for the output tape and the previously edited data input tape as well as keys to indicate whether edited data are present on the edited data input tape NIN and the scratch tape 12.

/MIXCOM/IMIX(4)

LOC	Symbol	Dimension	Description
1	LONNIN	<i>l</i>	Edited data present on tape NIN T: Yes F: No
2	LON12	<i>l</i>	Edited data present on tape 12 T: Yes F: No
3	NIN		Edited data tape input tape unit number. Initially = 11
4	NOUT		Edited data tape output tape unit number. Initially and terminally = 10

MIXCOM-1

/MLECOM/CMLE (1070)

Block data: Deck MC13B5

MLECOM is used for communication between the principal subroutines of the data start link of the DCP. It contains quantities describing the data to be used for the maximum likelihood estimation, option switches set during initialization, and temporary storage of the state and associated data.

/MLECOM/CMLE(1070)

Table 1

LOC	Symbol	Dimension	Definition
1	TSTART		Time from ONTIME to start of data start interval.
2	TFINAL		Time from ONTIME to end of data start interval.
3	TMEAN	d	Data start epoch, seconds from ONTIME (ST). Set to the mean time of the data to be used.
5	STIMR1	d	Seconds from 1950 Jan. 0.0 ST at TMEAN.
7	UTIMR1	d	UT at ST = STIMR1
9	ETIMR1	d	ET at ST = STIMR1
11	ETIMV1	d	ET at vehicle retransmission of the signal received at STIMR1
13	STIMR	d	ST of data point being processed.
15	UTIMR	d	UT at STIMR
17	UTIMR	d	ET at STIMR
19	ETIMV	d	ET at vehicle retransmission of signal received at STIMR.
21	UTIMT	d	UT at transmission by the transmitting station.
23	ETIMT	d	ET at UTIMT
25	DELAY	d	Down-leg transmission delay, ETIMR-ETIMV

MLECOM-1

Table 1 (Continued)

LOC	Symbol	Dimension	Definition
27	STBIAS	d(2)	Receiving station clock bias and bias rate, STIMR-UTIMR at UTIMR = 0 and $\frac{d}{dt}$ (STIMR-UTIMR).
31	OBIAS	d(4)	Measurement biases.
39	DATAT	d(4)	Measurement values at STIMR.
47	Q	d(4)	Variances of measurement random errors.
55	QI	d(4)	$QI(I) = 1/Q(I)$
63	TD2C	d(3, 3)	Transform from true equator, equinox of date at ETIMR1 to C-frame.
81	TB12C	d(3, 3)	Transform from B- to C-frame at UTIMR1.
99	X1	d(6)	Vehicle state with respect to Earth, C-frame, at ETIMV1.
111	XA1	d(6)	Vehicle state with respect to the central body, C-frame, at ETIMV1.
123	XM	d(6)	Vehicle state in measurement coordinates at ETIMV1.
135	PX12M	d(6, 6)	$\partial X1/\partial XM$
207	YMOD	d(4)	Modulus of measurement values.
215	YR	d(4)	Measurement residuals, $y-\hat{y}$.
223	PI	d(6, 6)	Information matrix, $H^T Q^{-1} H$.
295	HY	d(6)	Projected residuals, $H^T Q^{-1} (y-\hat{y})$.
307	TEST	d(20)	Vector of residual sums (see Table 2).
347	COEF	d(2, 4)	Coefficients of measurement polynomials.
363	AMU	d	Central body gravitational constant.
365	CTAU	d	Constant doppler interval or count, from the edited data tape.
367	DRA	d	Nutation in right ascension at ETIMR1.
369	Spares	(12)	-----
381	ILMEAS	l(4, 4)	Measurement inclusion options versus measurement number, type (T if measurement is included).
397	LMEAS	l(4)	Measurement options for this type, this process.

MLECOM-2

Table 1 (Continued)

LOC	Symbol	Dimension	Definition
401	LDATA	$l(4)$	Measurement options for this time point.
405	LAUTO	l	Internal automatic retry option (see KAUTO).
406	LMLE	l	Maximum likelihood convergence indicator, T if converged.
407	LRESDL	l	Residual switch, T output only, F include in information matrix and HY.
408	Spares	(3)	-----
411	INROT	(3, 4)	Rotation indices for each measurement type (see MEAS2X).
423	NROT	(4)	Rotation indices for current measurement type.
427	NYR	(4)	Number of measurements included in statistics.
431	NSTATE		Number of state elements being estimated.
432	NPTSTT		Number of time points loaded in DATA.
433	IQUAL		Quality indicator value for full set of measurements.
434	ITIME		Column counter of current time in DATA.
435	ILBM	(12, 4)	BCI measurement labels.
483	ILBS	(12, 4)	BCI measurement coordinate labels.
531	LBM	(12)	} labels for this measurement type.
543	LBS	(12)	
555	LBALG	(2)	BCI antenna alignment label.
557	LBDPL		BCI label for 3-way doppler.
558	LBTAU	(2)	BCI label for doppler mode.
560	LBTYP	(4)	BCI measurement type labels.
564	Spares	(7)	-----
571	DATA	(5, 100)	Interval data buffer, DATA (1, I) = time tag for Ith data "point." DATA (M+1, I) = Mth measurement value.

TEST d(20)

Table 2

Component	Symbol	Definition
1-4	TEST(M)	Sum of the squares of the residual for each measurement.
5	E	Weighted sum of the squares of the residuals $\sum QI(M)*TEST(M)$.
6	EBAR	Mean weighted sum of the squares, $\sum QI(M)*TEST(M)/NYR(M)$.
7	EDEL	Fractional change in E, $(E-ELAST)/E$.
8	EHAT	Value of E predicted for next trial.
9	EDHAT	Value of EDEL predicted for next trial.
10	Spare	-----
11-16	TEST(I+10)	Values of TEST(I+4) (E, etc.) on previous trial.
17	EHATX	Value of EHAT without constraint on the differential correction.
18	EDHATX	Value of EDHAT without constraint.
19-20	Spares	-----

MLECOM-4

```

SIBFTC MC13B5 M94,NODD,LIST
BLOCK DATA
C MAIN COMMON FOR DATA START LINK MLEC0001
C COMMON /MLECOM/CMLE(1070) MLEC0002
C DOUBLE PRECISION YMOD (4) MLEC0003
LOGICAL ILMEAS(4,4) MLEC0004
DIMENSION INROT(3,4) MLEC0005
1 ,ILBM (12,4) ,LBALG (2) ,LBTP (4) MLEC0006
2 ,ILBS (12,4) ,LBTAU (2) MLEC0007
C EQUIVALENCE (CMLE(435),ILBM ) ,(CMLE(557),LBDPL ) MLEC0008
1 ,(CMLE(483),ILBS ) ,(CMLE(558),LBTAU ) MLEC0009
2 ,(CMLE(411),INROT ) ,(CMLE(560),LBTP ) MLEC0010
3 ,(CMLE(381),ILMEAS) ,(CMLE(555),LBALG ) ,(CMLE(207),YMOD ) MLEC0011
C MEASUREMENT AMBIGUITIES MLEC0012
C DATA YMOD / 2*6.2831853071795864 ,2*1.D30 / MLEC0013
C MEASUREMENT INCLUSION KEYS MLEC0014
C DATA ILMEAS/ T,T,T,F ,T,T,T,T ,T,T,T,T ,T,T,F,T / MLEC0015
C ROTATION INDICES (FOR MEAS2X) MLEC0016
C DATA INROT / 1,2,3 ,-3,-2,1 ,-3,1,2 ,1,-2,3 / MLEC0017
C OUTPUT LABELS MLEC0018
C DATA ILRM /6HAZIMUT,6HELEVAT,6HRANGE ,6H ,6HH ,6HION MLEC0019
1 ,6H ,6H ,6HRAD ,6HRAD ,6HKM ,6H MLEC0020
2 ,6HX ,6HY ,6H2-WAY ,6HDOPPLE,6H ,6H MLEC0021
3 ,6HRANGE ,6HR ,6HRAD ,6HRAD ,6HSEC ,6HCYCLES MLEC0022
4 ,6HX ,6HY ,6H2-WAY ,6HDOPPLE,6H ,6H MLEC0023
5 ,6HRANGE ,6HR ,6HRAD ,6HRAD ,6HKM ,6HCYCLES MLEC0024
6 ,6HHOUR A,6HDECLIN,6H ,6HDOPPLE,6HNGLE ,6HATION MLEC0025
7 ,6H ,6HR ,6HRAD ,6HRAD ,6H ,6HCYCLES /MLEC0026
C DATA ILRS /6HRANGE ,6HAZIMUT,6HELEVAT,6HR-RATE,6HAZ-RAT,6HEL-RAT MLEC0027
1 ,6H ,6HH ,6HION ,6H ,6HE ,6HE MLEC0028
2 ,6HRANGE ,6HX ,6HY ,6HR-RATE,6HX-RATE,6HY-RATE MLEC0029
3 ,6HRANGE ,6HX ,6HY ,6HR-RATE,6HX-RATE,6HY-RATE MLEC0030
4 ,6H ,6H ,6H ,6H ,6H ,6H MLEC0031
5 ,6H ,6H ,6H ,6H ,6H ,6H MLEC0032
6 ,6HRANGE ,6HR.A. ,6HDECLIN,6HR-RATE,6HR.A.-R,6HDEC-RA MLEC0033
7 ,6H ,6H ,6HATION ,6H ,6HATE ,6HTE /MLEC0034
C DATA LBALG/6HN-S ,6HE-W / MLEC0035
DATA LBDPL/6HR-3WAY/ MLEC0036
DATA LBTAU/6HINTRVL,6HCOUNT / MLEC0037
DATA LBTP/6HC-BAND,6HG-RR ,6HUSBS ,6HDSIF / MLEC0038
C END MLEC0039

```

/OUTCOM/COU (40)

No block data.

OUTCOM is used by nearly every subroutine in the Tracking Data Editor. It is used for the transmission of output edited data to tape. It is written as the first record of each record pair on the edited data tapes. It contains key information about the associated data record written from MESCOM. It is also written without an associated data record as the last record on an edited data tape, in which case the end of tape condition is signalled by a flag in one of the words of the record.

/OUTCOM/COU (40)
Data Identification Buffer

LOC	Symbol	Dimension	Description
1	NREC		Record pair counter
2	NEOT		End of tape key -1, First record pair +1, Last record 0, Other
3	OBSNAM		Receiving station name
4	TRANAM		Transmitting station name
5	NRCD		Record pair counter for current arc
6	NPTS		Number of time points in data record
7	KONT		0; First record pair for this arc 1; Continuation record pair
8	MTYPE		Measurement system type 1; C-Band 2; Goddard Range & Range Rate 3; Unified S-Band 4; DSIF (JPL)
9	NALIGN		Antenna size (for USBS only) 1: 30-foot, principal axis north 2: 85-foot, principal axis east
10	MODE		Doppler mode 2: Two-way 3: Three-way
11	DELTA		Time interval between time points, seconds

OUTCOM-1

Data Identification Buffer (continued)

LOC	Symbol	Dimension	Description
12	KTAU		Doppler type: 0: Doppler count is fixed in COUT (19), data contains time. 1: Doppler interval is fixed in COUT (19), data contains counts.
13	ONTIME	d	Start of data arc, seconds from 1950
15	TFIRST	d	Start of record, seconds from 1950
17	TLAST	d	End of record, seconds from 1950
19	TAU	d	Fixed doppler count or interval (sec) see KTAU, COUT (12)
21	FTR	d	Transmitter carrier freq. Hz
23	CRI	d	} Refraction constants
25	CR2	d	
27	DR	d	
29	BIAS	d	Range ambiguity, units of range
29	BIAS	d	Fixed doppler offset bias, Hz
31	RETR	d	Transponder retransmission/reception ratio, e.g.: = 240/221 for USBS
33	NB	(4)	Outlier or missing data point count, four observables
37	SD	(4)	Sample standard deviations, four observables, units of observables, computed over entire arc.

/POTCOM/PLOT(42)

No block data.

POTCOM is used to communicate data to the plot routine RSPLIT. It is also used by RSPLIT to store quantities which must be saved if linking within the residual link.

/POTCOM/PLOT(42)

Loc	Symbol	Dimension	Description
1	TIME		Time (from initialization time) of present RSPLIT call, in seconds.
2	DATA	(4)	Up to 4 data values to be plotted at this time.
6	KDATA	(4)	If KDATA(I) = 1, plot DATA (I)
10	DT		Time scale increment, in seconds.
11	DATMAX	(4)	DATMAX (I) is the maximum value of DATA (I) to be plotted on scale.
15	LABEL	(12)	Each data type is identified on the plot by a one-character BCI symbol. The symbol is identified as to data type by preliminary output of a label which may be up to 12 characters in length. For data type I, I=1, 2, , 4 LABEL (I) is the 1st word of the BCI label. LABEL (I+4) is the 2nd word of the BCI label. LABEL(I+8) is the BCI label of units of DATA (I).
27	SYMBOL	(4)	and SYMBOL(I) is the BCI character used in the plot.
31	DORD	(4)	DORD (I) is used in placing the symbol for DATA(I) in the correct plot position.
35	TM	(3)	TM(1) = even time interval point of the last entry to RSPLIT. TM(2) = TM(1) - DT/2. TM(3) = TM(1) + DT/2.

POTCOM-1

38	LINE		Counter for the number of intervals (DT), to control output of a horizontal line every 10 intervals.
39	LINE2		Counter for the number of intervals, to control output of a time coordinate every 5 lines.
40	LINE3		Counter for the vertical axis label.
41	LTM		Temporary storage for a character of the vertical axis label.
42	N1		Used in assigned GO TO which controls output of vertical axis label.

POTCOM-2

/RSCOM/KPTIME(8), RLSAVE(56)

Block data: Deck MC135B

RSCOM contains the maximum values of the scales on which data is to be plotted. It is used by RESOUT in setting the data maximums for RSPLIT.

TABLE 1
/RSCOM/KPTIME(8)

Loc	Symbol	Dimension	Description
1	KPTIME		Logical variable if KPTIME is TRUE, the time interval for the plots is taken as PTIME. Otherwise, it is computed from BUFDAT.
2	PTIME		See KPTIME
3	KPMAX		If KPMAX is TRUE, the maximum value of data type M to be plotted on scale is taken as PMAX(M+1). Otherwise, it is computed to be PMAX(1) + SE(M+10, INDSTA).
4	PMAX	5	See KPMAX

RSCOM-1

TABLE 2
 RLSAVE (56)
 RESOUT Initialization Data

Loc	Symbol	Dimension	Definition
1	OBIAS	d(4)	Measurement biases
9	YMOD	d(4)	Measurement moduli
17	TFIRST	d	First station on-time, seconds from 1950 Jan 0.0
19	STPTIM		Process stop time, seconds from TFIRST
20	TREF		Start time for current data arc, seconds from TFIRST
21	Spare		-----
22	KOUTL	ℓ	Logical variable for inclusion of outliers
23	LSKIP	ℓ(4)	Measurement skip options
27	NSUM	(4)	Number of measurements included in SUM, SUMSQ
31	SUM	(4)	Sum of residuals, each type
35	SUMSQ	(4)	Sum of squares of residuals, each type
39	NASGN	(18)	Permanent transfer addresses (for assigned "GO TO's")

RSCOM-2

\$IBFTC MC135B LIST,NODD

```
BLOCK DATA
COMMON      /RSCOM/KPTIM(8),RLSAVE(56)
LOGICAL     KPMAX ,KPTIME
EQUIVALENC (KPTIM( 1),KPTIME) ,(KPTIM( 2),PTIME )
1           ,(KPTIM( 3),KPMAX ) ,(KPTIM( 4),PMAX )
DATA KPMAX,KPTIME/T,T/
DATA PTIME/60./
DATA PMAX/1.,0.1,0.1,2*100./
END
```

RSCM0001
RSCM0002
RSCM0003
RSCM0004
RSCM0005
RSCM0006
RSCM0007
RSCM0008

/RSLCOM/INDRSL(16), BUFRSL(18, 14)

No block data.

RSLCOM is used to keep track of output quantities related to the data and residuals. This data must be saved until the entire residual process is completed since the plotting, if requested, is output line by line for each time point. All other printer output must therefore be deferred until the plots are finished.

TABLE 1
INDRSL(16)

Loc	Symbol	Dimension	Description
1	NRSPOS		Present position on the residual tape (11)
2	NREOT		Record-type descriptor for the residual tape.
3	NRKONT		Data arc descriptor (=0 for a record not the last record of the arc) for the residual tape.
4	NRBLK		Number of INDRSL, BUFRSL, pairs written on residual tape.
5	NRPTS		Number of time points used.
6	NRSSTA		Station name for these residuals.
7	NREST		Number of estimation tape record pair used to define state for this residual process.
8	MTYPE		Data type. MTYPE = 1 C-Band 2 Goddard R, R 3 United S-Band 4 DSIF
9	TRON	d	Time to start computing residuals seconds from 1950 Jan 0, 0 ST
11	TROFF	d	Time to stop computing residuals seconds from 1950 Jan 0, 0 ST. If TROFF=0, this time is the end of the data arc.

RSLCOM-1

13	TRFRST	d	Time of the 1st residual in BUFRSL (in a block of 14 points on the tape), seconds from 1950 Jan 0.0 ST.
15	TRLAST	d	Time of the last residual of the block of 14, seconds from 1950 Jan 0.0 ST.

TABLE 2
BUFRSL(18, 14)

The residuals are collected in BUFRSL until 14 have been accumulated. They are put on tape (11) either as temporary storage before output or because the tape option was requested. For each point I, I=1, 14 the following is a table of BUFRSL(I, I)

Loc	Dimension	Definition
1		Quality word. This is a fixed point number, $= \sum 10^m$ for all M for which a valid residual appears for data type M.
2		Time of the data, seconds from 1950 Jan 0.0 ST.
3		Residual, data type 1
4		Residual, data type 2
5		Residual, data type 3
6		Residual, data type 4
7	d	X
9	d	Y
11	d	Z
13	d	\dot{X} State at time BUFRSL(2, I)
15	d	\dot{Y}
17	d	\dot{Z}

RLSCOM-2

/SBFCOM/CSBF(18N+12)

No block data.

SBFCOM is used for communication between the propagation and estimation links . It contains the quantities computed during the trajectory integration which are required in the processing of data.

The dimension of the arrays in SBFCOM depend on M, the maximum permissible number of motion parameters in the state vector. The dimension, N, is

$$N = 27 + 3M$$

The dimensions of the arrays and their location in CSBF are given as functions of N in the table below. The value of M used must be compiled into MAXQ of DCPCOM (see also DQDCOM). Changes in M require the recompilation of the following subroutines which reference SBFCOM.

<u>Subroutine</u>	<u>Deckname</u>
DIFCOR	MC133H
TRAJDP	MC133F
OUTTRJ	MC1327
ESTMAT	MC133J
ESTOUT	MC1330
UPDATP	MC138A

SBFCOM is loaded in the DIFCOR link of the DCP. It is loaded in the RSIDUL link under the name SB3COM, with fixed dimensions RBF(6), VBF (6, 8).

SBFCOM-1

/SBFCOM/CSBF (18N+12)

Location	General		Array	Definition
	Dimension	Loc		
1		1	KBF	Ordinate counter. Process type index. = 1: propagate = 2: establish anchor point = 3: estimate/shift anchor point
2		2	NTP	
3		3	NPT1	Data point counter value at iteration start. Data point counter value at iteration end.
4		4	NPT2	
5		5	LCB	Current central body.
6		6	ITER	Iteration counter.
7		7	HBF	Time increment for interpolation formulas.
9	d	9	TBF	Interpolation interval initial time.
11	d	11	TFF	Interpolation interval final time, TFF = TBF + 7 HBF.
13	d(N)	13	RBF	Initial "position" vector.
2N+13	d(N, 8)	103	VBF	"Velocity" ordinates, interpolation coefficients.

SBFCOM-2

/SB3COM/CSBF (120)

No block data.

SB3COM is used for communication between the sub-links of the RSIDUL link of the DCP. It contains the quantities computed during trajectory integration which are required for the computation of residuals. It is identical to SBFCOM except that the dimensions of RBF, VBF do not depend on the number of equation of motion parameters. For SB3COM, $N=6$ always.

SB3COM-1

/STNCOM/CSTN(7250)

Block Data: Deck MC1377

STNCOM is used for the storage of nominal station parameters, their standard deviations and correlation coefficients. It is used by SETSTA (only) in constructing the working arrays for the stations for which data is to be processed.

STNCOM may be changed only by recompilation of the block data. The working array, however, may be changed by overlay input at the case level (see References 1 and 2).

The dimensions of the arrays in STNCOM allow inclusion of up to 50 stations in the block data. If a larger station library is required and space is available, STNCOM may be redimensioned by replacing each "50" on card SSTA0041 by the new dimension.

/STNCOM/CSTN(7250)
Nominal Station Parameters

Loc	Symbol	Dimension	Description
1	STANAM	(50)	Station names
51	STALOC	d(3, 50)	Station locations Latitude (degrees) Longitude (degrees) Altitude (meters)
351	STACOR	d(69, 50)	Station location and measurement bias errors, standard deviations, and normalized correlations.

STNCOM-1

```

SIBFTC MC1377 XR3,M94,NODD,LIST
BLOCK DATA
TRACKING STATION BLOCK DATA
COMMON /STNCOM/CSTN(7250)
DIMENSION STANAM (50)
DOUBLE PRECISION STALOC(3,50),STACOR(69,50)
EQUIVALENCE (CSTN( 1),STANAM), (CSTN( 51),STALOC)
1 (CSTN(351),STACOR)
STATION NAMES
DATA STANAM,6HMERRIT,6HBAHAMA,6HANTIGU,6HBERMUD,6HCANARY
1,6HASCENS,6HCRNRVN,6HGUAM,6HHAWAII,6HGUAYMS
2,6HCORPUS,6HMADRID,6HMADRIX,6HCANBRA,6HCANBRX
3,6HGLDSTN,6HGLDSTX,6HVANGRD,6HREDSTN,6HMERCRY
4,6HHNTSVL,6HWTRTWN,6HKENEDY,6HPATRIK,6HSALVDR
5,6HGRTRUK,6HPRETOR,6HARGUEL,6HWSANDS,6HELGIN
6,6HJOBURG,6HWOMERA,6HROSAN,6HMADGAR,6HCARVON
7,6HSNTAGR,6HULASKR
8
9,13*6HSPARES
STATION LOCATIONS
LATITUDES (DEGREES)
DATA (STALOC(1,I),I=1,50)
1/ 28.42486D0, 26.63636D0, 17.14380D0, 32.34777D0, 27.74486D0
2, -7.97299D0, -24.90639D0, 13.58333D0, 22.12527D0, 27.95841D0
3, 27.65539D0, 40.41667D0, 40.41667D0, -35.31153D0, -35.31153D0
4, 35.38964D0, 35.38964D0, 0. D0, 0. D0, 0. D0
5, 0. D0, 0. D0, 28.48177D0, 28.22655D0, 24.11899D0
6, 21.46291D0, -25.94372D0, 34.58290D0, 32.35822D0, 30.42177D0
7, -25.88735D0, -31.38287D0, 35.20014D0, -19.02028D0, -24.87500D0
8, -33.14948D0, 64.97683D0,
9
1 13*0.D0/
LONGITUDES (DEGREES)
DATA (STALOC(2,I),I=1,50)
1/ -80.66442D0, -78.26772D0, 17.14380D0, -64.65364D0, -15.60200D0
2, -14.40169D0, 113.72369D0, 144.92500D0, -159.66762D0, -110.72079D0
3, -95.36358D0, -3.66667D0, -3.66667D0, 149.13583D0, 149.13583D0
4, -116.84878D0, -116.84878D0, 0. D0, 0. D0, 0. D0
5, 0. D0, 0. D0, -80.57651D0, -80.59929D0, -74.52074D0
6, -71.13204D0, 28.35850D0, -120.56115D0, -106.36956D0, -86.79811D0
7, 27.64878D0, 136.88502D0, 271.12790D0, 47.30278D0, 113.70833D0
8, 289.33091D0, 212.48490D0,
9
1 13*0.D0/
ALTITUDES (METERS)
DATA (STALOC(3,I),I=1,50)
1/ 9. D0, 13. D0, 26. D0, 3. D0, 36. D0
2, 143. D0, 58. D0, 20. D0, 1142. D0, 18. D0
3, 50. D0, 50. D0, 50. D0, 50. D0, 50. D0
4, 1031. D0, 1031. D0, 0. D0, 0. D0, 0. D0
5, 0. D0, 0. D0, 14. D0, 15. D0, 3. D0
6, 25. D0, 1618. D0, 646. D0, 1232. D0, 28. D0
7, 1381.92D0, 150.8 D0, 880.87D0, 1392.02D0, 75.01D0
8, 694.94D0, 294.44D0,
9
1 13*0.D0/
STATION ERRORS
NORTHING (KM)
DATA (STACOR( 1,I),I=1,50)/ 50* 0.D0 /
EASTING (KM)
DATA (STACOR( 2,I),I=1,50)/ 50* 0.D0 /
DOWN (KM)
DATA (STACOR( 3,I),I=1,50)/ 50* 0.D0 /
CLOCK BIAS (SEC)
DATA (STACOR( 4,I),I=1,50)/ 50* 0.D0 /
CLOCK RATE (SEC/SEC)
DATA (STACOR( 5,I),I=1,50)/ 50* 0.D0 /
SPDLT (KM/SEC)
DATA (STACOR( 6,I),I=1,50)/ 50* 0.D0 /
ANGLE 1 (RAD)
DATA (STACOR( 7,I),I=1,50)/ 50* 0.D0 /

```

C		ANGLE 2 (RAD)		STNC0075
	DATA (STACOR(8,I),I=1,50)/		50* 0.D0 /	STNC0076
C		RANGE (KM)		STNC0077
	DATA (STACOR(9,I),I=1,50)/		50* 0.D0 /	STNC0078
C		DOPPLER (COUNTS)		STNC0079
	DATA (STACOR(10,I),I=1,50)/		50* 0.D0 /	STNC0080
C	STANDARD DEVIATIONS			STNC0081
C		NORTHING (KM)		STNC0082
	DATA (STACOR(11,I),I=1,50)/		50*10.D-3 /	STNC0083
C		EASTING (KM)		STNC0084
	DATA (STACOR(12,I),I=1,50)/		50*10.D-3 /	STNC0085
C		DOWN (KM)		STNC0086
	DATA (STACOR(13,I),I=1,50)/		50*10.D-3 /	STNC0087
C		CLOCK BIAS (SEC)		STNC0088
	DATA (STACOR(14,I),I=1,50)/		50* 1.D-3 /	STNC0089
C		CLOCK RATE (SEC/SEC)		STNC0090
	DATA (STACOR(15,I),I=1,50)/		50* 1.D-12/	STNC0091
C		SPDLT (KM/SEC)		STNC0092
	DATA (STACOR(16,I),I=1,50)/		50*10.D0 /	STNC0093
C		ANGLE 1 (RAD)		STNC0094
	DATA (STACOR(17,I),I=1,50)/		50* 1.D-3 /	STNC0095
C		ANGLE 2 (RAD)		STNC0096
	DATA (STACOR(18,I),I=1,50)/		50* 1.D-3 /	STNC0097
C		RANGE (KM)		STNC0098
	DATA (STACOR(19,I),I=1,50)/		50* 1.D-1 /	STNC0099
C		DOPPLER (COUNTS)		STNC0100
	DATA (STACOR(20,I),I=1,50)/		50* 1.D0 /	STNC0101
C	CORRELATIONS			STNC0102
	DATA ((STACOR(I,J),J=1,50),I=21,65)/		2250* 0.D0 /	STNC0103
C	STANDARD DEVIATIONS IN MEAS RANDOM ERRORS (RAD,RAD,KM,COUNTS)			STNC0104
	DATA ((STACOR(I,J),J=1,50),I=66,69)/		100* 2.D-3 ,	STNC0105
	1		50* .3D-1 ,	STNC0106
	2		50*20.D0 /	STNC0107
	END			STNC0108
				STNC0109
				STNC0110

/STPCOM/CSTP(58)

No Block Data.

STPCOM is used for the storage of constants computed by the initialization subroutine STEPDI and the functions of the regularized incremental eccentric anomaly used in the solution of Kepler's equation by STEPDT.

/STPCOM/CSTP(58)

Conic Step Functions

Loc	Symbol	Dimension	Definition
1	TI	d	Initial time, seconds ET
3	XI	d(6)	Initial position and velocity, R_o, V_o
15	SU	d	$\sqrt{\mu}$, μ = gravitational constant
17	RI	d	$r_o = R_o $
19	AI	d	$1/a$, $ a $ = semi-major axis of the conic
21	SA	d	$\sqrt{1/ a }$
23	DI	d	$d_o = R_o \cdot V_o / \sqrt{\mu}$
25	SR	d	$\sqrt{\mu}/r_o$
27	PR	d	Orbital period (for ellipses)
29	HP	d	PR/2
31	OM	d	$(1 - r_o/a)$
33	EP	d	Convergence limit for STEPDT
35	R	d	$r = R $
37	G	d(4)	Universal functions, G_o, G_1, G_2, G_3
45	B	d	Regularized anomaly, β
47	FB	d	$F(\beta) = G_3 + d_o G_2 + r_o G_1$
49	D	d	$d = d_o G_o + (1 - r_o/a) G_1$
51	F	d(4)	R, V coefficients, f, g, \dot{f} , \dot{g}

STPCOM-1

/SUMCOM/SUMARY (56)

No block data.

SUMCOM is used to contain summary information about the output edited data tape. It is initially loaded by the MAIN program and is kept current by the decoding subroutines CBTEST, GRTEST, SRTEST, and DSTEST. It is also used by subroutines EGRESS and SCANIT. It is written as the first record on the output edited data tape.

/SUMCOM/SUMARY (56)

Data Tape Header Record

Loc	Symbol	Dimension	Description
1	HEADER	(11)	BCD tape header
12	NUMSTA		Total number of stations on tape
13	NAMSTA	(20)	Names of all stations on tape
33	TSTART	d	Time of first data on tape or earlier, seconds from 1950
35	TSTOP	d	Time of last data on tape or later, seconds from 1950
37	STIMNX	(20)	Time of earliest appearance of each station on tape as a receiving station (or earlier), seconds from 1950

SUMCOM-1

/TRJCOM/CTRJ(246)

No block data.

TRJCOM is used for the working storage of quantities required to be saved or communicated during trajectory computation. It is loaded in the DIFCOR link, and is relabelled TRSCOM for the MLESTT link and TR3COM for the RSIDUL link.

/TRJCOM/CTRJ(246)

Loc	Symbol	Dimension	Definition
1	ETFMS	d	ET of last call of DPFMRS
3	UTIMV	d	Current universal time, seconds from 1950 Jan 0.0 UT
5	TC2D0	d(3, 3)	Transformation, mean equator, equinox of 1950.0 (C), to mean equator, equinox of date (D0)
23	TC2D	d(3, 3)	Transformation, to true equator, equinox of date (D)
41	TB2D	d(3, 3)	Transformation, central body-fixed (B), to true equator, equinox of date
59	TB2C	d(3, 3)	Transformation, central body-fixed to mean equator, equinox of 1950.0
77	CBOD	d(37)	Central body working storage
151	RC	d(3)	Vehicle position relative to the central body, C-frame components
157	VC	d(3)	Vehicle velocity relative to the central body, C-frame components
163	RCONIC	d(3)	Encke reference position
169	VCONIC	d(3)	Encke reference velocity
175	PCONIC	d(6, 6)	Conic transition matrix

TRJCOM-1

CBOD d (37)
Central Body Working Storage

Location		Symbol	Dimension	Definition
CTRJ	CBOD			
77	1	CBN		Central body name
79	2	CBP	d	Sphere of influence
81	3	CBA	d	Equatorial radius
83	4	CBB	d	Polar radius
85	5	CBU	d	Gravitational constant
87	6	CBO	d	Angular velocity
89	7	DRA	d	Nutation in right ascension (Earth) of 1950.0 right ascension of the prime meridian (planets)
91	8	HRA	d	Hour angle of the vernal equinox
93	9	OBL	d	Mean obliquity of the ecliptic (Earth) or inclination of Earth's ecliptic on the planet's equator
95	10	DLN	d	Nutation in longitude (Earth) or 1950.0 longitude of the ascending node of the ecliptic on the equator (planets)
97	11	OBT	d	True obliquity (Earth) or Earth's mean obliquity of 1950.0 (planets)
99	12	NZH		Highest order zonal harmonic
101	13	NTH		Highest degree tesseral harmonic
103	14	ZH	d(6)	Zonal harmonic coefficients
115	20	TH	d(18)	Tesseral harmonic coefficients

TRJCOM-2

/TRKCOM/CTRK(700)

Block data: Deck MC134S

TRKCOM is used for storage of tracking station descriptive data. It is loaded by a Block Data subroutine, but may be altered by overlay input at the beginning of the program run.

/TRKCOM/CTRK (700)
Tracking Station Identification

Loc	Symbol	Dimension	Description
1	STANAM	(50)	Station names
51	KODSTA	(50)	Station ID codes
101	MOUNT	(50)	USBS antenna size 1, 30-foot, principal axis north 2, 85-foot, principal axis east
151	TRF	d(50)	Transmitter frequency Hz
251	C1	d(50)	} Refraction constants
351	C2	d(50)	
451	BIAS	d(50)	Doppler fixed offset bias, Hz, used only for USBS
551	RETR	d(50)	Transponder retransmission/reception ratio. Used for USBS and for Goddard Range - Range Rate in S-Band mode.
651	PAIR	(50)	Associated transmitting station names.

Use of the various arrays by the TDEP

STANAM

Tracking stations are identified on the Edited Data Tape by six-letter names rather than by numbers. This is the array of possible names.

TRKCOM-1

KODSTA

Tracking stations are identified on the Raw Data Tapes by two-digit decimal numbers. The TDEP reads the number from the raw data tape, searches for a match in KODSTA, picks up the corresponding name in STANAM, and puts this name on the edited data tape. Exception: Subroutine GRTEST (Goddard System Data) reads the number from the raw data tape, reassigns a new number, and then uses the reassigned value to search for a match in KODSTA. This is necessary to avoid conflicts due to duplication of station ID numbers. The reassignments are:

<u>On Tape</u>	<u>Reassigned</u>
26	30
22	31
52	32
27	33
28	34

Care must be exercised if subroutine GRTEST is to be altered to change either the list of acceptable numbers from the raw data or the reassigned values. Reassigned values less than 33 indicate to subroutine GRTEST that the tracker is a regular system, whereas numbers greater than or equal to 33 indicate a hybrid system tracker.

NALIGN

This array has meaning only for USBS data and is used only by subroutine SBTEST.

FTR

This array is meaningless for C-Band data and is ignored by CBTEST. It is also ignored by GRTEST, which contains its own list of possible carrier frequencies.

TRKCOM-2

C1 and C2

These are used by all data subroutines.

BIAS

This array is meaningless for C-Band data and is ignored by CBTEST. It is also ignored by GRTEST, which contains its own list of bias frequencies.

RETR

This array is meaningless for C-Band data and is ignored by CBTEST. It is always used by SBTEST. It is used by GRTEST for S-Band frequencies (but not VHF) but the Differential Correction Program (Subroutines DSDAT and DSTATP) were altered to ignore this value for Goddard data because the Goddard System receivers shift the received signal back up by the inverse of the transponder ratio to give an effective value of unity. This fact became apparent after the TDEP was written.

PAIR

This array is used only for USBS data in the three-way doppler mode in case the transmitting and receiving stations are not the same.

TRKCOM-3

```

$IBFTC MC134S M94,NODD,XR3
CMC134S STATION BLOCK DATA FOR TRACKING DATA EDITOR
BLOCK DATA
COMMON /TRKCOM/CTRK(700)
DOUBLE PRECISION FTR(50), C1(50), C2(50), BIAS(50), RETR(50)
DIMENSION STANAM(50), KODSTA(50), NALIGN(50), PAIR(50)
EQUIVALENCE (CTRK(1),STANAM), (CTRK(51),KODSTA)
1, (CTRK(101),NALIGN), (CTRK(151),FTR), (CTRK(251),C1)
2, (CTRK(351),C2), (CTRK(451),BIAS), (CTRK(551),RETR)
3, (CTRK(651),PAIR)
C
DATA (STANAM(I),KODSTA(I),NALIGN(I),PAIR(I),FTR(I),C1(I),C2(I)
1,BIAS(I),RETR(I),I=1,9)/
C**TRACKING SHIPS
C VANGUARD
16HVANGRD, 62, 1, 6HVANGRD, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C REDSTONE
16HREDSTN, 63, 1, 6HREDSTN, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C MERCURY
16HMERCRY, 64, 1, 6HMERCRY, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C HUNTSVILLE
16HHNTSVL, 65, 1, 6HHNTSVL, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C WATERTOWN
16HWTRTWN, 66, 1, 6HWTRTWN, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C**LAND STATIONS
C MERRITT ISLAND
16HMERRIT, 71, 1, 6HMERRIT, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C GRAND BAHAMA ISLAND
16HBAHAMA, 41, 1, 6HBAHAMA, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C ANTIGUA
16HANTIGU, 91, 1, 6HANTIGU, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C BERMUDA
16HBERMUD, 02, 1, 6HBERMUD, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286/
DATA (STANAM(I),KODSTA(I),NALIGN(I),PAIR(I),FTR(I),C1(I),C2(I)
1,BIAS(I),RETR(I),I=10,18)/
C CANARY ISLANDS
16HCANARY, 04, 1, 6HCANARY, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C ASCENSION
16HASCENS, 75, 1, 6HASCENS, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C CARNARVON
16HCRNRVN, 08, 1, 6HCRNRVN, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C GUAM
16HGUAM, 24, 1, 6HGUAM, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C HAWAII
16HHAWAII, 12, 1, 6HHAWAII, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C GUAYMAS
16HGUAYMS, 14, 1, 6HGUAYMS, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C CORPUS CHRISTI
16HCORPUS, 16, 1, 6HCORPUS, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C MADRID
16HMADRID, 23, 2, 6HMADRID, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286,
C MADRID (BACKUP)
16HMADRIX, 94, 2, 6HMADRIX, 2045.69D6, .34D-3, .138771D-3
2, 1.D6, 1.085972859728597286/
DATA (STANAM(I),KODSTA(I),NALIGN(I),PAIR(I),FTR(I),C1(I),C2(I)
1,BIAS(I),RETR(I),I=19,22)/
C CANBERRA
16HCANBRA, 25, 2, 6HCANBRA, 2045.69D6, .34D-3, .138771D-3

```

	2,	1.D6,		1.085972859728597286,				CTRK0075
C	CANBERRA (BACKUP)							CTRK0076
	16HCANBRX,	93,	2,	6HCANBRX,	2045.69D6,	.34D-3,	.138771D-3	CTRK0077
	2,	1.D6,		1.085972859728597286,				CTRK0078
C	GOLDSTONE							CTRK0079
	16HGLDSTN,	28,	2,	6HGLDSTN,	2045.69D6,	.34D-3,	.138771D-3	CTRK0080
	2,	1.D6,		1.085972859728597286,				CTRK0081
C	GOLDSTONE (BACKUP)							CTRK0082
	16HGLDSTX,	92,	2,	6HGLDSTX,	2045.69D6,	.34D-3,	.138771D-3	CTRK0083
	2,	1.D6,		1.085972859728597286/				CTRK0084
	DATA (STANAM(I),KODSTA(I),NALIGN(I),PAIR(I),FTR(I),C1(I),C2(I)							CTRK0085
	1,BIAS(I),RETR(I),I=23,27)/							CTRK0086
	C**GODDARD RANGE-RANGE RATE SYSTEM STATIONS							CTRK0087
	C**REGULAR							CTRK0088
C	ROSMAN							CTRK0089
	16HROSAN,	30,	1,	6HROSAN,	2271.9328D6,	.34D-3,	.138771D-3	CTRK0090
	2,	.5D6,		.75D0,				CTRK0091
C	MADAGASCAR							CTRK0092
	16HMADGAR,	31,	1,	6HMADGAR,	2271.9328D6,	.34D-3,	.138771D-3	CTRK0093
	2,	.5D6,		.75D0,				CTRK0094
C	CARNARVON							CTRK0095
	16HCARVON,	32,	1,	6HCARVON,	2271.9328D6,	.34D-3,	.138771D-3	CTRK0096
	2,	.5D6,		.75D0,				CTRK0097
	C**HYBRID							CTRK0098
C	SANTIAGO							CTRK0099
	16HSNTAGR,	33,	1,	6HSNTAGR,	2271.9328D6,	.34D-3,	.138771D-3	CTRK0100
	2,	.5D6,		.75D0,				CTRK0101
C	ALASKA							CTRK0102
	16HULASKR,	34,	1,	6HULASKR,	2271.9328D6,	.34D-3,	.138771D-3	CTRK0103
	2,	.5D6,		.75D0/				CTRK0104
	END							CTRK0105

/TSTCOM/CTEST(400)

No block data.

TSTCOM is used for storage of a variety of program controls and constants. Some are obtained from input data; others are computed.

/TSTCOM/CTEST(400)

LOC	Symbol	Dimension	Description
1	NYR	(10)	Year numbers for raw data tapes
11	NPTS	(10)	Number of points to include in polynomial fit, each tape
21	NSTEP	(10)	Polynomial walk step size in points, each tape
31	NDEG	(10)	Degree of polynomial to fit, each tape
41	CSD	(4, 10)	Number of standard deviations allowable for outlier tagging. Each of 4 observables in following order: <ol style="list-style-type: none"> 1. Azimuth, X, or hour angle 2. Elevation, Y, or declination 3. Range 4. Doppler
81	IRAW		Current location in above arrays; sequence number of raw data tape in process
82	KNTROL	(4)	Use indicator for each observable, standard order: <ol style="list-style-type: none"> 0: no data, this observable 1: good data, this observable 4: Goddard system angles, valid data every fourth time point
86	(Spare)		-----
87	FINISH	d(10)	Time of last data to be processed, seconds from 1950, each tape.
107	IFOMIT	(4,10)	Data suppression keys, each observable in standard order, each tape. <ol style="list-style-type: none"> +1: suppress 0: do not suppress

TSTCOM-1

/TSTCOM/CTEST(400) (continued)

LOC	Symbol	Dimension	Description
147	KMAX	(10)	Number of valid station ID codes, each tape.
157	ISTAOK	(20, 10)	List of valid station codes, each tape uses one column.
357	TLO	d(10)	} Bounds for valid data time tags, seconds from 1950, each tape.
377	THI	d(10)	
397	Spares		
400			-----

TSTCOM-2

/WCOM/IW(550), CW(1450)

No block data.

WCOM is used for storage of a large number of working common arrays that must be made available to a number of subroutines. All the values are generated by the program and may not be altered by any data overlay process.

/WCOM/IW(550)

Table 1

LOC	Symbol	Dimension	Definition
1	NSTX	(3)	Extra body keys once used by SHIFT. Not now used by SHIF2.
4	NOX	(3)	Array of central body numbers for up to three extra bodies; e.g., if the first extra body is earth centered, NOX(1) = 1.
7	IXT		Total number of extra bodies
8	ISC	(12)	Upper-loaded array of numbers of stations to be considered. For example, ISC = (1, 5, 12, 0, ..., 0) indicates that only stations 1, 5, and 12 are to be considered.
20	IBC	(10)	Upper-loaded array of numbers of beacons to be considered.
30	IEMP	(24)	Not used
54	IOBR	(4)	Reserved for indicators for on-board measurements. Not presently in use.
58	IOBA	(18)	
59	Spares	(122)	

WCOM-1

Table 1 (Continued)

LOC	Symbol	Dimension	Definition
180	IS	(11, 12)	<p>The second index corresponds to the station identification number. The first row of this array IS(1, N) contains a 1, 2, 3, or 4 to indicate the type of measurements to be made as follows:</p> <ul style="list-style-type: none"> 1: C-Band 2: Goddard Range and Range Rate 3: Unified S-Band 4: DSIF (JPL) <p>The next nine rows are error source treatment indicators. Row order is:</p> <ul style="list-style-type: none"> 2 Range treatment 3 Doppler treatment 4 Angle 1 treatment 5 Angle 2 treatment 6 Northing error treatment 7 Easting error treatment 8 Down error treatment 9 Clock bias treatment <p>For rows 2-5 the following code is used:</p> <ul style="list-style-type: none"> 0: No measurement 1: Measure with random and bias errors 2: Measure with random errors only <p>Rows 6-9 use the following code:</p> <ul style="list-style-type: none"> 0: No error 1: Include bias error <p>Row 10 is a key to indicate antenna alignment (dish size) for USBS systems</p> <ul style="list-style-type: none"> 1: 30-foot dish, principal axis north 2: 85-foot dish, principal axis east <p>Row 11 is a key for refraction effects.</p> <ul style="list-style-type: none"> 1: Include 2: Omit

WCOM-2

Table 1 (Continued)

LOC	Symbol	Dimension	Definition
312	IB	(3, 12)	<p>Each of the first ten columns corresponds to a beacon number; the rows contain indicators for beacon location errors north, east, and down in that order. The code is identical to that used for rows 6-9 of the IS array. Columns 11 and 12 contain:</p> <p>IB(1, 11) Range treatment indicator IB(2, 11) Range-rate treatment indicator IB(3, 11) Angle 1 treatment indicator IB(1, 12) Angle 2 treatment indicator IB(2, 12) Time bias treatment indicator IB(3, 12) Body number on which all beacons are located</p> <p>The treatment indicators are like those used for station error treatments.</p>
347	NBEAC		NBEAC = IB(3, 12), the number of the body on which all beacons are located.
348	Spares	(7)	-----
355	ISEE	(12)	<p>Spacecraft in-view indicator for stations. Upper-loaded to correspond to ISC array</p> <p>-1: Cannot see spacecraft 0: Can see spacecraft</p>
367	IBEE	(10)	<p>In-view indicator for beacons. Upper-loaded to correspond to IBC array.</p> <p>-1: Cannot be seen by spacecraft 0: Can be seen by spacecraft</p>
377	ICAS		Case number used to search for desired trajectory on tape 10.
378	Spares	(2)	-----
380	NMAX		Number of columns in measurement buffer XMES (NMAX = 85)
381	Spare		-----
382	NOR		Current central body number.
383	ITARG		Target body number. Only the target body can occult the spacecraft as viewed from the earth.

WCOM-3

Table 1 (Continued)

LOC	Symbol	Dimension	Definition
384	IK1		Indicator showing current location in SSTART and SSTOP arrays.
385	KSMAX		Total number of station critical events found by SBEV2.
386	IK2		Indicator showing current location in BSTART and BSTOP arrays.
387	KBMAX		Total number of beacon critical events found by SBEV2.
388	Spare		-----
389	ISTIM	(50)	Event type indicator in conjunction with STIME array times: +K: Station K into view -K: Station K out of view 13: Occulting starts 14: Occulting stops
439	IBTIM	(50)	Event type indicator in conjunction with BTIME array times: +K: Beacon K into view -K: Beacon K out of view
489	ITRIG		Trigger to start record counter KOUNT in EXINST: 0: No count 1: Count
490	KOUNT		Record counter incremented in EXINST when tape 10 is read and ITRIG = 1. Used for rapid repositioning of tape after search for critical events.
491	Spares	(60)	-----

WCOM-4

/WCOM/CW(1450)

Table 2

LOC	Symbol	Dimensions	Definition
1	RX	(3, 3)	Extra bodies' initial position vectors, km, equator and equinox of 1950. RX(I, J) is the Ith coordinate of extra body J with respect to body NOX(J).
10	VX	(3, 3)	Extra bodies' initial velocity vectors, km/sec, arranged as above.
19	TX	(2, 3)	Initial dates for extra bodies Row 1 Whole days from 1950. 0 Row 2 Fractional days from 1950. 0
25	Spares	(131)	-----
156	BIASL	(3, 22)	The second index refers to station or beacon number; 1-12 for stations 1-12, 13-22 for beacons 1-10. Contains location errors in km, north, east, down in that order.
222	SSTART	(75)	Station on-times, seconds from epoch, arranged chronologically. Corresponding station numbers are in KSTA array.
297	SSTOP	(75)	Station off-times, seconds from epoch, arranged to pair with SSTART times.
372	BSTART	(75)	Beacon on-off times. Analogous to the SSTART and SSTOP arrays. Beacon numbers in KBEA array.
447	BSTOP	(75)	
522	XMES	(5, 85)	Storage array for measurements to be written on tape 12. Each column contains a set of measurements and a time tag. The rows are: 1 Time, seconds from epoch 2 Angle 1, radians 3 Angle 2, radians 4 Range, km or seconds 5 Doppler, counts This table is written on tape 12 by rows.
947	Spares	(187)	-----

WCOM-5

TABLE 2 (Continued)

LOC	Symbol	Dimension	Definition
1134	XIN	(6)	Current spacecraft state vector body NOR, equator and equinox of 1950, km and km/sec.
1140	TSEC		Current time from epoch in seconds.
1141	Spares	(16)	-----
1157	REFS	(3)	Not used - reserved for on-board measure- ment reference data.
1160	TI	(3)	Current date in days from 1950.0 1 Whole days 2 Fractional days 3 Not used
1163	AN	(3, 3)	Transformation from equator and equinox of 1950 to equator and equinox of date.
1172	TSECO		Initial time, seconds from epoch.
1173	G HAR		Greenwich hour angle at TSECO, radians.
1174	Spares	(4)	-----
1178	TSTART		Time, in seconds from epoch, at which measurements are to start or, if in process, time to read new control cards.
1179	FLTIM		Time in seconds from epoch at which to end the case.
1180	EVNT	(6)	Chronologically ordered critical event array which contains the next to be considered of each of the following six types of events: Start station measurements Start beacon measurements Start on-board optical measure- ments Start on-board radar measure- ments TSTART FLTIM
1186	Spares	(2)	-----

WCOM-6

TABLE 2 (Continued)

LOC	Symbol	Dimension	Definition
1188	TWT	(2)	Control times in seconds from 1950.0 1. Time to start measurements 2. Time to stop measurements and read next control cards.
1190	Spares	(10)	-----
1200	STIME	(50)	Ordered array of times at which critical events ISTIM occur.
1250	BTIME	(50)	Ordered array of times at which critical events IBTIM occur.
1360	SECR	(12)	Time array in calling sequence of SBEV2. Not now used.
1372	SHFC	(18)	Orbital parameters for extra bodies, set up by STEPI
1390	Spares	(36)	-----
1426	TWO		Initial time in whole days from 1950.0
1427	TFO		Initial time in fractional days from 1950.0
1428	Spares	(13)	-----
1441	BECR	(10)	Time array in calling sequence of SBEV2. Not now used.

WCOM-7

SECTION 2

SUBROUTINES

2.0 INTRODUCTION

This section contains the descriptions and listings of all subroutines and functions used by the Orbit Determination Program system, arranged in alphabetical order. Each subroutine description lists the functions of the subroutine and the rules for its usage, including required input and resulting output. All referenced commons and subroutines not contained in the standard IBSYS library are listed. Equations describing the computations are given where required, and references are quoted for necessary derivations.

Where labelled commons are referenced by the subroutines, a minimum description of the common variables is given. For more detailed descriptions, reference should be made to Section 1 of this report.

Blank common is used throughout the program system as a "scratch pad;" that is, for strictly internal temporary storage. Some care is required in any changes of subroutines which use blank common to avoid loss of stored variables. Normally, a call to a subroutine which uses N cells of blank common may be expected to destroy any data stored in these cells, and the scratch pad for the using subroutine should therefore begin at cell N+1. The length of blank common specified by the subroutine description is the length used by the subroutine and any referenced subroutines.

2.1 SUMMARY OF SUBROUTINE FUNCTIONS

The paragraphs below contain brief functional statements for each of the programs in the ODP system and their component subroutines. A tabular cross-reference summarizing all references to all commons and user-supplied subroutines is also given.

2.1.1 Differential Correction Program (DCP)

The Differential Correction Program accepts an a priori estimate of the space vehicle state and other parameters and a tape containing measurement values, and statistically combines these data into an improved estimate of the state and parameters.

DCP SUBROUTINES

Table 2.1

Subroutine Name	Function
ACCTRJ	Computes second derivatives for integration of the equations of motion and variational equations.
ACCTR3	Computes second derivatives for integration of the equations of motion.
BLDCOV	Builds portions of the covariance matrix.
CBDAT	Computes C-band observables as functions of vehicle state.
CBDATP	Computes C-band observables and partial derivatives as functions of vehicle state.
COVOUT	Writes normalized covariance matrix on the system output tape.
DATINP	Converts date input to seconds from 1950 January 0.0.
DATOUP	Converts seconds from 1950 January 0.0 to date format and writes date on the system output tape.
DCP	Main program for the DCP.
DCROSS	Computes the cross product of two vectors.
DDOT	Computes the dot product of two vectors.
DEHA	Computes hour angle of the vernal equinox and Earth's rotation rate.

Table 2.1 (Continued)

Subroutine Name	Function
DEQD	Integrates equations of motion and variational equations.
DEQ3	Integrates equations of motion.
DEQTR	Computes parameters defining the precession of Earth's equator.
DGTRN	Computes transformation matrix from a defined sequence of rotations.
DGTSN	Computes transformation matrix from sines and cosines of rotation angles.
DIFCOR	Main program for propagation and differential correction.
DINVRT	Computes the inverse of an $N \times N$ matrix.
DLUNE	Computes the transformation from Earth's equator, equinox of date to Moon-fixed coordinates.
DMPY	Computes the product of two matrices.
DMVTRN	Computes the product of a 3×3 matrix and a $3 \times N$ matrix.
DNORM	Computes the magnitude of a 3-vector.
DPRMRS	Computes planetary positions and velocities.
DRAGD	Computes atmospheric drag.
DRAGDP	Computes atmospheric drag and its partial derivatives.
DSDAT	Computes DSIF observables from vehicle state.
DSDATP	Computes DSIF observables and their partial derivatives from vehicle state.
DSTAT	Computes tracking station position vectors and transformation matrices.
DTRANP	Transforms covariance matrix from base coordinates to output coordinates.

Table 2.1 (Continued)

Subroutine Name	Function
DTRDB	Computes transformation matrix from base coordinates to Darboux coordinates.
DVNORM	Computes the magnitude of a 3-vector, and normalizes the vector.
ENCKED	Computes Encke acceleration.
ESTMAT	Computes and adds differential corrections to the state and covariance matrix.
ESTOUT	Writes state estimate and covariance matrix on the estimate tape and on the system output tape.
GRAVD	Computes accelerations due to gravitational harmonics.
GRAVDP	Computes accelerations and partial derivatives for gravitational harmonics.
GRDAT	Computes Goddard Range Range Rate observables from vehicle state.
GRDATP	Computes Goddard observables and partial derivatives from vehicle state.
GTR2BD	Computes central body transformations.
MEAS2X	Computes cartesian coordinates from measurement coordinates.
MLESTT	Main program for maximum likelihood data start.
MXLEST	Computes maximum likelihood estimate of vehicle state.
OUTTRJ	Writes integration output and forms state interpolation tables for differential correction.
OUTTR3	Forms state interpolation tables for residual computation.
OUTXPD	Writes vehicle state on system output tape.
OVRLYD	Double precision input subroutine.

TABLE 2.1 (Continued)

Subroutine Name	Function
PERTD	Computes planetary gravitational accelerations.
PERTDP	Computes planetary gravitational accelerations and their partial derivatives.
PSEUDO	Computes the pseudo inverse of an $N \times N$ matrix.
RESOUT	Controls output of residuals on residual tape and system output tape.
RSPLOT	Plots residuals on the system output tape.
SBDAT	Computes S-band observables from vehicle state.
SBDATP	Computes S-band observables and partial derivatives from vehicle state.
SETCAS	Performs case initialization. Builds <u>a priori</u> state and covariance.
SETSTA	Sets up working arrays of station data.
SETTAP	Checks or writes tape headers and initializes tape positions.
SOLRD	Computes solar pressure acceleration.
SOLRDP	Computes solar pressure acceleration and partial derivatives.
STATE	Computes vehicle state and transformation.
STEPDI	Initializes conic step functions.
STEPDP	Computes conic transition matrix.
STEPDT	Computes position and velocity on a conic.
STTBGN	Initializes maximum likelihood start.
STTDAT	Sets up data arrays for the m. l. e. start.
STTEND	Stores completed maximum likelihood estimate.

Table 2.1 (Continued)

Subroutine Name	Function
STTFIT	Fits individual measurements to a time polynomial.
STTIME	Computes residuals for maximum likelihood start.
TRAJD	Initializes and controls trajectory integration for the residual computation.
TRAJDP	Initializes and controls trajectory integration for propagation and differential correction.
UPDATP	Updates the covariance matrix in time.
VENTD	Computes acceleration due to venting.
VENTDP	Computes acceleration due to venting and its partial derivatives.
X2ORBD	Computes orbital elements from cartesian coordinates.

2.1.2 Tracking Data Editing Program (TDEP)

The Tracking Data Editing Program accepts as input one or more BCD tapes of tracking data, each in one of four acceptable formats. The TDEP reads these data, performs necessary unit conversions, detects and tags outliers, and writes these data on a binary tape in a format suitable for input to the DCP. As an optional input, a tape of previously edited data may be combined with new data and written as the TDEP output tape.

TDEP SUBROUTINES

Table 2.2

Subroutine Name	Function
CBTEST	Reads and decodes a BCD tape of C-Band tracking data.
CROSIM	Solves a set of simultaneous, linear algebraic equations.
DATINP	Converts date input to seconds from 1950 January 0.0.
DAToup	Converts seconds from 1950 January 0.0 to date format and writes date on the system output tape.
DSTEST	Reads and decodes a BCD tape of DSIF (JPL) tracking data.
EGRESS	Transfers the final edited data to tape 10 if necessary and writes a summary of the edited data on the system output tape.
GRTEST	Reads and decodes a BCD tape of Goddard Range and Range Rate System tracking data.
MIXIT	Reads edited data from tapes 12 and NIN; combines these data and writes them on tape NOUT.
NUDATA	Allows the nominal values in the tracking station arrays to be overlaid with different values.
• OPTW.	Allows an optional return from the IBSYS error subroutine FXEM when illegal characters are encountered on tracking data read-in.
POLYFT	Tests a data arc (station pass) of edited data with walking polynomials; tags outliers.
SBTEST	Reads and decodes a BCD tape of unified S-band tracking data.
SCANIT	Reads the final edited data tape and writes all the data on the system output tape.
TDEP	Main program for Tracking Data Editing Program; performs overall control.

2.1.3 Tracking Data Simulator (TDS)

The Tracking Data Simulator accepts as input a spacecraft trajectory tape, a planetary ephemeris tape, and one or more sets of input cards that specify a tracking station network and the time interval over which measurements are to be made. The TDS uses these inputs to write a tape of simulated measurement data in a format suitable for input to the Differential Correction Program.

TDS SUBROUTINES

Table 2.3

Subroutine Name	Function
ANTR1	Provides planetary positions and velocities by interpolation on values read from the planetary ephemeris tape.
ARKTNS	Four-quadrant arctangent subroutine.
BACK	Backspaces a specified tape (always tape 10 in the TDS) a specified number of records.
BARN	Random number generator. Will generate uniform or Gaussian random variables.
BECSTA	Computes pairs of on and off times for stations and beacons.
BIBCD	Converts a binary integer to BCD.
BMEAS	Makes a set of beacon measurements at a specified point in time.
CARDIN	Reads data cards that specify the tracking network or changes thereto.
CRITA	Outputs the time at which a body starts or stops occulting the vehicle.
CRITIC	Takes arrays of station- or beacon-related critical events and forms pairs of on-off times.

Table 2.3 (Continued)

Subroutine Name	Function
CRITO	Outputs the station or beacon critical events.
CROSS	Computes the cross product of two vectors.
DATOUT	Converts whole and fractional days from 1950 January 0.0 to date and writes date on the system output tape.
DELEV	Sorts the critical event array into chronological order and determines the required measurement interval.
DOT	Computes the scalar product of two vectors.
ERROUT	A FORTRAN dummy replacement for the MAP-coded error trace routine of the same name.
EXINST	Provides spacecraft position and velocity by interpolation on tables read from the spacecraft trajectory tape.
FIEF	Reads tapes forward or backspaces tapes over end-of-file marks.
FIFL	Finds a specified case on the spacecraft trajectory tape and reads the first record thereof.
FIST	Rewinds the spacecraft trajectory tape and checks the header. Also initializes FIFL.
FNORM	Computes the magnitude of a vector.
GHA	Computes the Greenwich hour angle of the first point of Aries at a given time.
GOTOR	Solves Kepler's equation for incremental eccentric anomaly given incremental mean anomaly.
INITAP	Initialization subroutine for the TDS.
LAYO	Reads station, (beacon and onboard) data cards and converts units as required.
LOCAT	Adds location errors to station and beacon locations.

Table 2.3 (Continued)

Subroutine Name	Function
MEASUR	Computes and stores an arc of appropriate type measurement data.
MESERP	Directs the overall logic flow for making simulated data tapes with the TDS.
MNA	Provides the rotation matrix that transforms coordinates in the earth's true equator and equinox to coordinates in the moon's true equator.
MTRN	Computes the 3 x 3 product of two 3 x 3 matrices.
NUTAIT	Computes the transformation from mean equator and equinox to true equator and equinox.
ORB	Computes orbital elements from a Cartesian state vector and writes the elements in the system output tape.
OYAL	Writes the data read by LAYO on the system output tape.
PARAB	Fits a parabola through three points.
QUARTC	Finds the zeros of a quadratic.
ROTEQ	Computes the transformation from mean equator and equinox of 1950 to mean equator and equinox of date.
ROVLEY	Reads fixed, floating, and alphanumeric data into labeled common.
SBEV2	Creates an ordered array of times at which the vehicle comes into or goes out of view of stations or beacons.
SHIF2	Given the state of a body or spacecraft with respect to a body, returns the state with respect to any other body or bodies.
SKDOUT	Writes out the measurement schedule on the system output tape.
SKDUL	Reads the control time cards and calls SKDOUT to output the measurement schedule.

Table 2.3 (Continued)

Subroutine Name	Function
SMEAS	Makes a set of station measurements of a specified type at a given time.
SORDR	Sorts an array X into ascending order, preserving the correspondence between X and another array NX.
SORDR2	Sorts an array X into ascending order, preserving the correspondence between X and another array NX and, optionally, a third array Y.
STARTB	Loads the next beacon on time from the BSTART array into position EVNT(1) of the critical event array.
STARTS	Loads the next station on time from the SSTART array into position EVNT(1) of the critical event array.
STAT	Obtains a vector from the body center to a station or beacon and computes an orthogonal transform relating inertial Cartesian quantities X, Y, Z to local tangent plane quantities North, East, Down.
STEPI	Performs initialization for STEPT
STEPT	Finds the state of a body on a conic at a time, that conic having been initialized by a call of STEPI.
TCONIC	Computes the time from periapsis corresponding to a given true anomaly on a given conic section.
TDS	Executive driver for the TDS.
TFRAC	Computes integral and fractional parts of a sum. Used to update the internal date format.
TIMED	Converts a time interval from time format to seconds.
TIMES	Converts a time interval in seconds to time format and prepares a six-word BCD array for simplified output of interval in days, hours, minutes, and seconds.
VNORM	Normalizes a vector.
VTRN	Given a 3x3 matrix A and a 3x1 vector X, computes the vector AX.
VTRT	Given a 3x3 matrix A and a 3x1 vector X, computes the vector $A^T X$.

2.1.4 Residual Output Program (ROP)

The Residual Output Program accepts as input the estimate and residual tapes written by the DCP for use in special purpose analysis of the residuals. The program is intended only as a guide for the use of the input tapes, and each analytical function required must be programmed as required. Only the main program is supplied.

2.2 SUBROUTINE DESCRIPTIONS

Descriptions and listings of each of the subroutines used by the ODP are given in the following pages. Each subroutine is written as a separate section to facilitate replacement as the subroutines are modified.

Subroutine: ACCTRJ

Purpose: To compute the perturbation acceleration for Encke integration of the equations of motion. Also, computes derivatives for variational equations.

Calling Sequence: CALL ACCTRJ

Common storages used: //144 cells, /DCPCOM/, /DFMCOM/, /DQDCOM/, /ESTCOM/,
/TRJCOM/

Subroutines required: DCR0SS, DD0T, DMPY, DMVTRN, DPFMRS, DRAGDP, DVN0RM,
ENCKED, GRAVDP, GTR2BD, PERTDP, STEPDP, STEPDT, VENTDP,
S0LRDP

ACCTRJ-1

Usage:

ACCTRJ accepts the integrated dependent variables RDEQ,VDEQ of /DQDCOM/, and computes the accelerations ADEQ. The components of RDEQ,VDEQ,ADEQ are

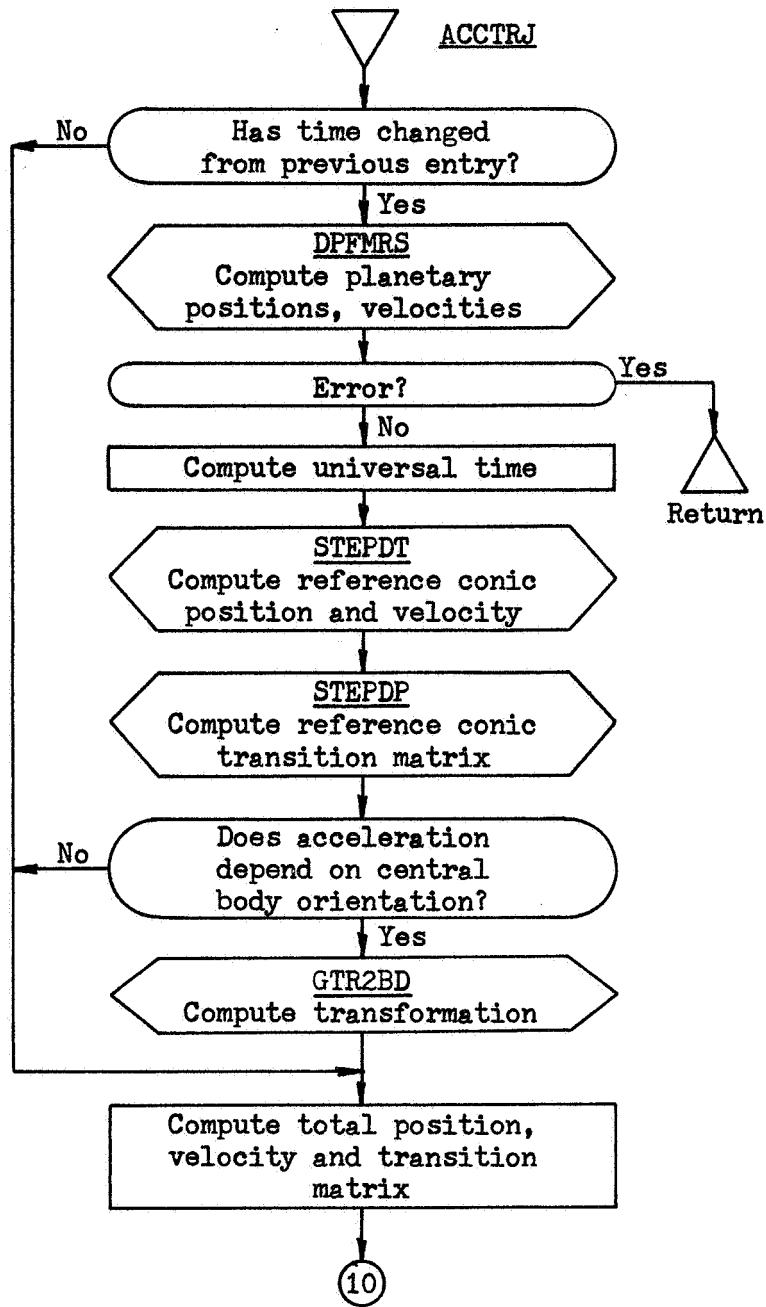
I = 1, 2, 3 R , \dot{R} , \ddot{R} , in C-frame coordinates

I = 4 - 21 φ , $\dot{\varphi}$, $\ddot{\varphi}$

I = 22 - (NPEND-1) φ_u , $\dot{\varphi}_u$, $\ddot{\varphi}_u$

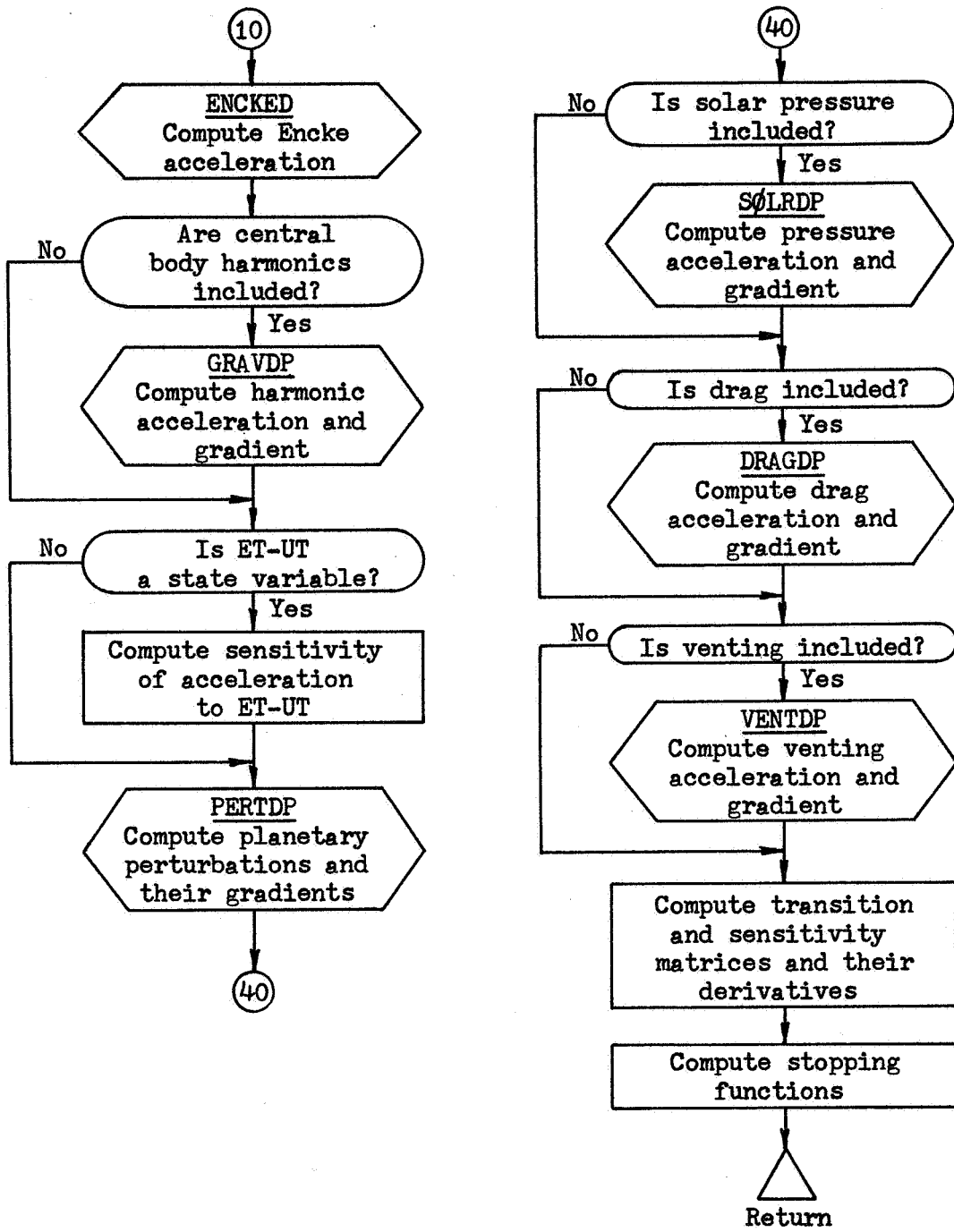
The equation of motion model is described in Appendix D of Reference 1.

ACCTRJ-2



Central Body Position, Velocity and Orientation
 Vehicle Total Position, Velocity

ACCTRJ-3



Accumulation of
Perturbation Accelerations

ACCTRJ-4

```

SIBFTC MC133A XR3,M94,NODD,LIST
SUBROUTINE ACCTRJ
C COMPUTES PERTURBATION ACCELERATION FOR ENCKE INTEGRATION
C COMPUTES STATE TRANSITION MATRIX DERIVATIVE AND
C EQUATION OF MOTION PARAMETER SENSITIVITIES
C REFERENCE - TR-DA1508, APPENDIX D
C
C DOUBLE PRECISION DDOT ,DSQRT ,DVNORM
C
C COMMON /DCPCOM/CDCP(900)
DOUBLE PRECISION DIRECT ,RSTP2
DIMENSION CBODY(8,11) ,ETAPE (4) ,IFEMP(8)
1 ,IPL (11) ,IHM (24) ,NEMP (8)
EQUIVALENCE
1 (CDCP(112),ETAPE ) ,(CDCP(663),IPL )
2 (CDCP(777),ICB ) ,(CDCP(779),KCB )
3 (CDCP(17),CBODY ) ,(CDCP(687),IFEMP ) ,(CDCP(695),NEMP )
4 (CDCP(809),DIRECT ) ,(CDCP(703),IHM ) ,(CDCP(897),RSTP2 )
EQUIVALENCE
1 (IFEMP( 8),IFBODY) ,(NEMP ( 1),NDRAG )
2 (IFEMP( 7),IFDRAG) ,(NEMP ( 8),NEMPS )
3 (IFEMP( 6),IFHARM) ,(NEMP ( 7),NEMP3 )
4 (IFEMP( 3),IFSOLR) ,(NEMP ( 2),NHARM )
5 (IFEMP( 5),IFTIME) ,(NEMP ( 6),NPEND )
6 (IFEMP( 4),IFVENT) ,(NEMP ( 3),NSOLR )
7 (NEMP ( 5),NTIME )
C
C COMMON /ESTCOM/CEST(804)
DOUBLE PRECISION DELDAN(2) ,EFEDAN(14) ,EHADAN(24,3) ,PREDAN(4)
1 ,ETIMVA
EQUIVALENCE
1 (CEST( 97),DELDAN) ,(CEST(101),EHADAN)
2 (CEST( 61),EFEDAN) ,(CEST( 89),PREDAN)
3 (CEST( 29),ETIMVA)
C
C COMMON /TRJCOM/CTRJ(246)
DOUBLE PRECISION CBOD(37) ,ETFMS ,PCONIC(6,6) ,RC(3) ,TB2C(3,3)
1 ,CBO ,CBU ,UTIMV ,RCONIC(6) ,VC(3)
2 ,CBP ,VCONIC(3)
EQUIVALENCE
1 (CTRJ( 1),ETFMS ) ,(CTRJ( 59),TB2C )
2 (CTRJ(175),PCONIC) ,(CTRJ( 3),UTIMV )
3 (CTRJ(151),RC ) ,(CTRJ(157),VC )
4 (CTRJ( 77),CBOD ) ,(CTRJ(163),RCONIC) ,(CTRJ(169),VCONIC)
EQUIVALENCE
1 (CBOD ( 3),CBA ) ,(CBOD ( 5),CBU ) ,(CBOD (20),TH )
2 (CBOD ( 6),CBO ) ,(CBOD (13),NTH ) ,(CBOD (14),ZH )
C
C COMMON /DFMCOM/IFM(14),RFM(6,12)
1 ,DFM (4) ,BFM (577) ,SNT(2,102)
DOUBLE PRECISION BFM,DFM,RFM
C
C COMMON /DQDCOM/NDEQ(10) ,CDEQ(10) ,XDEQ (4)
1 ,ADEQ(44) ,RDEQ(44) ,VDEQ(44) ,FDEQ(44,10)
DOUBLE PRECISION ADEQ,FDEQ,RDEQ,VDEQ,XDEQ
DOUBLE PRECISION ETIMV
EQUIVALENCE
1 (XDEQ,ETIMV)
C
C COMMON D ,DTP ,GRAD
DOUBLE PRECISION D(48) ,DTP(3,2) ,GRAD(3,6)
C
C POSITION AND VELOCITY
C
C 1 IF (ETFMS.EQ.ETIMV) GO TO 2
PLANETARY POSITIONS AND VELOCITIES
CALL DPFMRS (ETIMV,EFEDAN,ICB,IERR,ETAPE)
IF (IERR.NE.0) GO TO 999
ETFMS = ETIMV
UTIMV = ETIMV-DELDAN(1)-DELDAN(2)*ETIMV
C REFERENCE CONIC
CALL STEPDT (ETIMV,RCONIC)
CALL STEPDP (PCONIC)
IF (IFBODY.EQ.0) GO TO 2
C CENTRAL BODY ORIENTATION
CALL GTR2BD (ICB,DFM,J)
C POSITION AND VELOCITY RELATIVE TO CENTRAL BODY
2 CONTINUE

```

	DO 3 I=1,3	ACCJ0075
	RC(I) = RDEQ(I)+RCONIC(I)	ACCJ0076
	VC(I) = VDEQ(I)+VCONIC(I)	ACCJ0077
	DO 3 J=1,6	ACCJ0078
	GRAD(I,J) = 0.00	ACCJ0079
C	3 CONTINUE	ACCJ0080
C	ENCKE ACCELERATION	ACCJ0081
C		ACCJ0082
C	10 CALL ENCKED (CBU,RCONIC,RDEQ,ADEQ)	ACCJ0083
C		ACCJ0084
C	GRAVITATIONAL HARMONICS	ACCJ0085
C		ACCJ0086
C	20 IF (IFHARM.EQ.0) GO TO 30	ACCJ0087
	CALL GRAVDP (CBA,CBU,NZH,NTH,ZH,TH,RC,TB2C,ADEQ,IFHARM	ACCJ0088
	1 ,GRAD,ADEQ(NHARM),IHM,IHM(7))	ACCJ0089
	IF (IFTIME.EQ.0) GO TO 30	ACCJ0090
C	ET-UT SENSITIVITIES	ACCJ0091
	K = -1-IFTIME	ACCJ0092
	CALL DCROSS (RC,TB2C(1,3),DTP)	ACCJ0093
	DO 21 I=1,3	ACCJ0094
	21 DTP(I,1) = CBO*DTP(I,1)	ACCJ0095
	IF (K.EQ.0) GO TO 23	ACCJ0096
	DTP(3,2) = ETIMV-ETIMVA	ACCJ0097
	DO 22 I=1,3	ACCJ0098
	22 DTP(I,K) = DTP(3,2)*DTP(I,1)	ACCJ0099
	GO TO 24	ACCJ0100
	23 K = 1	ACCJ0101
	24 CALL DMVTRN (GRAD,DTP,ADEQ(NTIME),1,K)	ACCJ0102
C		ACCJ0103
C	PLANETARY PERTURBATIONS	ACCJ0104
C		ACCJ0105
C	30 CALL PERTDP (ICB,IPL,RC,EFEDAN,RFM,ADEQ,ADEQ(22),GRAD,1)	ACCJ0106
C		ACCJ0107
C	SOLAR PRESSURE	ACCJ0108
C		ACCJ0109
C	40 IF (IFSOLR.EQ.0) GO TO 50	ACCJ0110
	CALL SOLRDP (RC,RFM(1,10),CBA,PREDAN,ADEQ,ADEQ(NSOLR),GRAD,IFSOLR)	ACCJ0111
C		ACCJ0112
C	ATMOSPHERIC DRAG	ACCJ0113
C		ACCJ0114
C	50 IF (IFDRAG.EQ.0) GO TO 60	ACCJ0115
	CALL DRAGDP (RC,VC,TB2C,CBO,CBA,PREDAN(2),ADEQ	ACCJ0116
	1 ,ADEQ(NDRAG),GRAD,IFDRAG)	ACCJ0117
		ACCJ0118
C		ACCJ0119
C	VENTING	ACCJ0120
C		ACCJ0121
C	60 IF (IFVENT.EQ.0) GO TO 80	ACCJ0122
	CALL VENTDP (VC,PREDAN(4),ADEQ,ADEQ(NVENT),GRAD(1,4),IFVENT)	ACCJ0123
C		ACCJ0124
C	EMP SENSITIVITIES	ACCJ0125
C		ACCJ0126
C	80 IF (NEMPS.EQ.0) GO TO 83	ACCJ0127
	K = 22	ACCJ0128
	DO 82 I=1,4,3	ACCJ0129
	CALL DMVTRN (GRAD(1,I),RDEQ(K),D(1),NEMPS)	ACCJ0130
	DO 81 J=1,NEMPS	ACCJ0131
	81 ADEQ(J+22) = ADEQ(J+22)+D(J)	ACCJ0132
	82 K = K+MAXD	ACCJ0133
C		ACCJ0134
C	TRANSITION MATRIX	ACCJ0135
	83 CALL DMVTRN (GRAD(1,1),RDEQ(4),D(1),1,6)	ACCJ0136
	CALL DMVTRN (GRAD(1,4),VDEQ(4),D(19),1,6)	ACCJ0137
	D(40) = CBU/DVNORM(RCONIC,D(37))**3	ACCJ0138
	D(41) = 3.00*D(40)	ACCJ0139
	DO 84 I=1,3	ACCJ0140
	D(42) = D(41)*D(I+36)	ACCJ0141
	GRAD(I,I) = GRAD(I,I)+D(40)	ACCJ0142
	DO 84 J=1,3	ACCJ0143
	84 GRAD(I,J) = GRAD(I,J)-D(42)*D(J+36)	ACCJ0144
	CALL DMPLY (GRAD,PCONIC,ADEQ(4),3,6,6,3,6,3,0)	ACCJ0145
	DO 85 I=1,18	ACCJ0146
	85 ADEQ(I+3) = ADEQ(I+3)+D(I)+D(I+18)	ACCJ0147
C		ACCJ0148
C	STOPPING FUNCTIONS	ACCJ0149

```

C
90 IF (NDEQ(3).EQ.0) GO TO 999
   D(1) = DDOT(RC,RC)
   ADEQ(NPEND+1) = D(1)-CBP*CBP
   DO 91 I=1,3
91  D(I+1) = RC(I)-RFM(I,KCB)
   ADEQ(NPEND ) = DDOT(RDEQ,RDEQ)/D(1)-DRECT*DRECT
   ADEQ(NPEND+2) = DDOT(D(2),D(2))-CRODY(2,KCB)*CBODY(2,KCB)
   ADEQ(NPEND+3) = D(1)-RSTP2
   ADEQ(NPEND+4) = DDOT(RC,VC)
999 RETURN
   END

```

```

ACCJ0150
ACCJ0151
ACCJ0152
ACCJ0153
ACCJ0154
ACCJ0155
ACCJ0156
ACCJ0157
ACCJ0158
ACCJ0159
ACCJ0160

```

Subroutine: ACCTR3

Purpose: To compute the perturbation acceleration for Encke
integration of the equations of motion. (See also ACCTRJ).

Calling Sequence: CALL ACCTR3

Common storages used: //90 cells, /DCPCOM/, /DFMCOM/, /DQ3COM/, /ESRCOM/,
/TRJCOM/

Subroutines required: DDOT, DPFMRS, DRAGD, ENCKED, GRAVD, GTR2BD, PERTD,
STEPDT, VENTD, SOLRD

ACCTR3-1

```

SIBFTC MC133B XR3,M94,NODD,LIST
SUBROUTINE ACCTR3
C COMPUTES PERTURBATION ACCELERATION FOR ENCKE INTEGRATION
C REFERENCE - TR-DA1508, APPENDIX D
C
C DOUBLE PRECISION DDOT
C
COMMON /DCPCOM/CDPC(900)
DOUBLE PRECISION DRECT ,RSTP2
DIMENSION CBODY(8,11) ,ETAPE (4) ,IFEMP(8)
EQUIVALENCE
1 (CDPC(809),DRECT ) ,(CDPC(687),IFEMP ) ACC30010
2 (CDPC(112),ETAPE ) ,(CDPC(663),IPL ) ACC30011
3 (CDPC(777),ICB ) ,(CDPC(779),KCB ) ACC30012
4 (CDPC(17),CBODY ) ,(CDPC(111),IERR ) ,(CDPC(897),RSTP2 ) ACC30013
EQUIVALENCE (IFEMP(7),IFDRAG) ,(IFEMP(3),IFSOLR) ACC30014
1 (IFEMP(8),IFBODY) ,(IFEMP(6),IFHARM) ,(IFEMP(4),IFVENT) ACC30015
C
COMMON /ESRCOM/CESR(304)
DOUBLE PRECISION DELDAR(2) ,EFEDAR(14) ,EHADAR(24,3) ,PREDAR(4)
1 ,ETIMVA
EQUIVALENCE (CESR(97),DELDAR) ,(CESR(101),EHADAR)
1 (CESR(61),EFEDAR) ,(CESR(89),PREDAR)
2 (CESR(29),ETIMVA)
C
COMMON /TRJCOM/CTRJ(246)
DOUBLE PRECISION CBOD(37) ,ETFMS ,RCONIC(3) ,RC(3)
1 ,CBO ,CBP ,UTIMV ,VCONIC(3) ,VC(3)
EQUIVALENCE (CTRJ(151),RC ) ,(CTRJ(3),UTIMV )
1 (CTRJ(77),CBOD ) ,(CTRJ(163),RCONIC) ,(CTRJ(157),VC )
2 (CTRJ(1),ETFMS ) ,(CTRJ(59),TB2C ) ,(CTRJ(169),VCONIC)
EQUIVALENCE (CBOD(2),CBP ) ,(CBPD(12),NZH )
1 (CBOD(3),CBA ) ,(CROD(5),CBU ) ,(CBOD(20),TH )
2 (CBOD(6),CBO ) ,(CROD(13),NTH ) ,(CROD(14),ZH )
C
COMMON /DFMCOM/IFM(14),RFM(6,12)
1 ,DFM(4),BFM(577),SNT(2,102)
DOUBLE PRECISION BFM,DFM,RFM
C
COMMON /DQ3COM/NDEQ(10),CDEQ(10),XDEQ(4)
1 ,ADEQ(8),RDEQ(8),VDEQ(8),FDEQ(8,10)
DOUBLE PRECISION ADEQ,FDEQ,RDEQ,VDEQ,XDEQ
DOUBLE PRECISION ETIMV
EQUIVALENCE (XDEQ,ETIMV)
C
COMMON D
DOUBLE PRECISION D(4)
C
C POSITION AND VELOCITY
C
C
1 IF (ETFMS.EQ.ETIMV) GO TO 2
C PLANETARY POSITIONS AND VELOCITIES
CALL DPFMRS (ETIMV,EFEDAR,ICB,IERR,ETAPE)
IF (IERR.NE.0) GO TO 999
ETFMS = ETIMV
UTIMV = ETIMV-DELDAR(1)-DELDAR(2)*ETIMV
C REFERENCE CONIC
CALL STEPDT (ETIMV,RCONIC)
IF (IFBODY.EQ.0) GO TO 2
C CENTRAL BODY ORIENTATION
CALL GTR2BD (ICR,DFM,0)
C POSITION AND VELOCITY RELATIVE TO CENTRAL BODY
2 CONTINUE
DO 3 I=1,3
RC(I) = RDEQ(I)+RCONIC(I)
VC(I) = VDEQ(I)+VCONIC(I)
3 CONTINUE
C
C ENCKE ACCELERATION
C
10 CALL ENCKED (CBU,RCONIC,RDEQ,ADEQ)
C
C GRAVITATIONAL HARMONICS
C
20 IF (IFHARM.EQ.0) GO TO 30
CALL GRAVD (CBA,CBU,NZH,NTH,ZH,TH,RC,TB2C,ADEQ,IFHARM)

```

C		ACC30075
C	PLANETARY PERTURBATIONS	ACC30076
C		ACC30077
	30 CALL PERTD (ICR,IPL,RC,EFEDAR,RFM,ADEQ)	ACC30078
C		ACC30079
C	SOLAR PRESSURE	ACC30080
C		ACC30081
	40 IF (IFSOLR.EQ.0) GO TO 50	ACC30082
	CALL SOLRD (RC,RFM(1,10),CBA,PREDAR,ADEQ)	ACC30083
C		ACC30084
C	ATMOSPHERIC DRAG	ACC30085
C		ACC30086
	50 IF (IFDRAG.EQ.0) GO TO 60	ACC30087
	CALL DRAGD (RC,VC,TB2C,CBO,CBA,PREDAR(2),ADEQ)	ACC30088
C		ACC30089
C	VENTING	ACC30090
C		ACC30091
	60 IF (IFVNT.EQ.0) GO TO 90	ACC30092
	CALL VENTD (VC,PREDAR(4),ADEQ)	ACC30093
C		ACC30094
C	STOPPING FUNCTIONS	ACC30095
C		ACC30096
	90 IF (NDEQ(3).EQ.0) GO TO 999	ACC30097
	D(1) = DDOT(RC,RC)	ACC30098
	ADEQ(5) = D(1)-CBP*CBP	ACC30099
	DO 91 I=1,3	ACC30100
	91 D(I+1) = RC(I)-RFM(I,KCR)	ACC30101
	ADEQ(4) = DDOT(RDEQ,RDEQ)/D(1)-DRECT*DRECT	ACC30102
	ADEQ(6) = DDOT(D(2),D(2))-CBODY(2,KCB)*CBODY(2,KCB)	ACC30103
	ADEQ(7) = D(1)-RSTP2	ACC30104
	ADEQ(8) = DDOT(RC,VC)	ACC30105
	999 RETURN	ACC30106
	END	

Subroutine: ANTR,ANTR1

Purpose: To interpolate as a function of time for the coordinates of celestial bodies relative to a given central body.

Calling Sequence: CALL ANTR,ANTR1(TW,TF,NB,P ϕ ,NV,VE,DIS)

Input and Output

I/ ϕ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	TW				Time in whole days from 1950 Jan 0.0 UT.
I	TF				Time in fractional days from TW.
I	NB				Central body numbers.
ϕ	P ϕ	(3,7)		km	Array of position values.
I	NV				Dummy variable
ϕ	VE	(3,7)		km/sec	Array of velocity values.
I	DIS			km	Dummy variable.

Common storages used: None

Subroutines required: None

ANTR-1

Usage:

ANTR is a single-precision planetary ephemeris routine which uses a JPL tape of modified central differences. The bodies and their associated numbers are:

I	Body I
0	Earth
1	Moon
2	Sun
3	Venus
4	Mars
5	Saturn
6	Jupiter

The coordinates of all central bodies are computed relative to the body NB. The output arrays contain

$\phi(1-3,I)$ = Position of body I relative to
body NB, mean equator, equinox of 1950.0.
 $VE(1-3,I)$ = Velocity of body I relative to
body NB.

ANTR computes only the positions, and ANTR1 computes both positions and velocities.

Method

On each entry, ANTR locates and reads the proper tape record, containing

$$R_0 = R(T_j)$$

$$R_1 = R(T_j + h)$$

h = ephemeris interval

and the modified central differences $\delta^2 R_0$, $\delta^2 R_1$, $\delta^4 R_0$, $\delta^4 R_1$.

ANTR-2

The ephemeris time,

$$T = TW + TF + \Delta T$$

where

$$T_j \leq T < T_j + h$$

is meaningful, using

$$t = (T - T_j)/h$$

$$u = 1 - t$$

and the positions and velocities at time T are obtained from Everett's formula and its derivative

$$y(t) = \left\{ uy_0 + ty_1 \right\} + \left\{ \frac{u(u^2 - 1)}{3!} \delta^2 y_0 + \frac{t(t^2 - 1)}{3!} \delta^2 y_1 \right\} \\ + \left\{ \frac{u(u^2 - 1)(u^2 - 4)}{5!} \delta^4 y_0 + \frac{t(t^2 - 1)(t^2 - 4)}{5!} \delta^4 y_1 \right\}$$

$$\dot{y}(t) = \left\{ -y_0 + y_1 \right\} + \left\{ -\frac{3u^2 - 1}{3!} \delta^2 y_0 + \frac{3t^2 - 1}{3!} \delta^2 y_1 \right\} \\ + \left\{ -\frac{5u^4 - 15u^2 + 4}{5!} \delta^4 y_0 + \frac{5t^4 - 15t^2 + 4}{5!} \delta^4 y_1 \right\}$$

ANTR-3

\$IBMAP	MC13FM	600,MFTC		
*	ANTR			ANTR0010
	ENTRY	EWORTO		ANTR0020
	ENTRY	HWORTO		ANTR0030
	ENTRY	INTR		ANTR0040
	ENTRY	INTR1		ANTR0050
	ENTRY	ANTR		ANTR0060
	ENTRY	ANTR1		ANTR0070
	ENTRY	GRAV		ANTR0080
	ENTRY	SCALE1		ANTR0090
	FDH	OPSYN	FDP	ANTR0100
EGM	EQU	GRAV		ANTR0110
MGM	EQU	GRAV+1		ANTR0120
ANTR	NULL			ANTR0130
	INTR	TXL *+3,**	POSITION ENTRY	ANTR0140
ANTR1	NULL			ANTR0150
INTR1	NULL			ANTR0160
	SXD	VEL,4	POSITION-VELOCITY ENTRY	ANTR0170
	TXL	*+2,**		ANTR0180
	STZ	VEL		ANTR0190
	SXA	LDIR,4		ANTR0200
	SXA	SYSLOC,4		ANTR0210
	CLA	6,4		ANTR0220
	ADD	=21		ANTR0230
	STA	OUP		ANTR0240
	CLA	8,4		ANTR0250
	ADD	=21		ANTR0260
	STA	XN.V		ANTR0270
TAR	CLA*	3,4		ANTR0280
	STO	TARG		ANTR0290
FRA	CLA*	4,4		ANTR0300
	STO	TARG+1		ANTR0310
CEN	CLA*	5,4		ANTR0320
	STO	CENTER		ANTR0330
RADS	CLA*	9,4		ANTR0340
	STO	R		ANTR0350
	SXD	TRAP,4		ANTR0360
	SXD	TRAP+1,2		ANTR0370
	SXD	HELIO-1,1		ANTR0380
	ZET	FILE		ANTR0390
	TRA	NFIL		ANTR0400
	CALL	.FVIO.(TAPNO,FILE)		ANTR0410
	CLA	FILE		ANTR0420
	STA	FLE		ANTR0430
	STA	FEL		ANTR0440
	STA	FLEA		ANTR0450
	STA	FLPP		ANTR0460
	TSX	.OPEN,4		ANTR0470
FLPP	MON	*		ANTR0480
	CALL	SETN(COM,COM+1)		ANTR0490
	CALL	.FVIO.(COM+1,FLUT)		ANTR0500
NFIL	CLA	VEL		ANTR0510
	TNZ	NEU	MUST INTERPOLATE TO	ANTR0520
	REM		OBTAIN VELOCITY	ANTR0530
	CLA	CENTER		ANTR0540
	SUB	KERNO		ANTR0550
	TNZ	NEU	MUST INTERPOLATE FOR	ANTR0560
	REM		NEW CENTRAL BODY	ANTR0570
	CLA	TARG		ANTR0580
	SUB	TARGO		ANTR0590
	TNZ	NEU	TIME HAS CHANGED	ANTR0600
	CLA	TARG+1		ANTR0610
	SUB	TARGO+1		ANTR0620
	TZE	FLEE+6		ANTR0630
	REM		ANALYZE TABLE	ANTR0640
	REM		NEEDS AS A	ANTR0650
	REM		FUNCTION OF	ANTR0660
	REM		CENTRAL BODY	ANTR0670
NEU	CLA	TARG		ANTR0680
	FSB	TABLE		ANTR0690
	TMI	LOOK	POSITION EPHEMERIS TAPE	ANTR0700
	STO	COM+19	T-T0	ANTR0710
	SUB	20.		ANTR0720
	TPL	LOOK		ANTR0730
	CLA	COM+19		ANTR0740

FDH 4.			
STQ COM+18			
STQ TEM	(T-T0)/4		
*IFIX TEM			
STO TEM			
*FLOAT TEM			
STO COM+17			
CHS			
FAD COM+18			
STO COM+1			
CLA TARG+1			
FDH 4.			
STQ COM			
CLA COM			
FAD COM+1			
STO TARG+2			
LDQ COM+19			
FMP =9.			
STO TEM			
*IFIX TEM			
ADD KERN0+1			
STA GG			
LDQ COM+17			
FMP =63.	HELIO SPACING		
STO TEM			
*IFIX TEM			
ADD KERN0+2			
STA HH			
AXT 2*SEP,4	MUST INTERPOLATE		
STZ XN+2*SEP,4	SO CLEAR STORAGE		
TIX *-1,4,1			
AXT 6,4			
STZ SATPOS+6,4			
TIX *-1,4,1			
AXT BSEP,4	SET GRAVITATIONAL		
STZ KB0+BSEP,4	COEFFICIENTS		
TIX *-1,4,1	TO ZERO		
LXA CENTER,4			
PXD ,4			
COM			
PDX ,2			
TRA HELIO			
GEO AXT 3,1			
CLA GRAV+3,1			
STO KB0+3,1			
TIX *-2,1,1			
STZ KB0-1,2			
CLA EWORT0			
STO EWORT			
CLA HWORTE			
STO HWORT			
TXH GG12,4,0			
CLA RJ	TEST FOR		
SUB R	INCLUSION OF		
TPL GG12	JUPITER AND SATURN		
CLA HWORTJ	IF EARTH IS		
STO HWORT	CENTRAL BODY		
CLA GRAV+5			
STO KB6-1			
CLA GRAV+6			
STO KB6			
HELIO TXL GG12,***			
AXT BSEP,1			
CLA GRAV+BSEP,1			
STO KB0+BSEP,1			
TIX *-2,1,1			
STZ KB2-3,2			
ORTHO CLA EWORT0			
STO EWORT			
CLA HWORT0			
STO HWORT			
GG12 CLA VEL	CHECK		
TZE GG6	FOR VELOCITY		
CLA EWORT0	OPTIONS		
STO EWORT			

ANTR0750
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 ANTR1490

	CLA	HWORTV		ANTR1500
	STO	HWORT		ANTR1510
GG6	CLA	EWORT		ANTR1520
	TZE	GG+2		ANTR1530
	CLA	EVEL		ANTR1540
	STO	VEL+1		ANTR1550
	CLA	TARG+1	GEOCENTRIC	ANTR1560
	TSX	TAB,4	INTERPOLATION	ANTR1570
GG	PZE	**,,9		ANTR1580
	PZE	XN,,EWORT+1		ANTR1590
	CLA	HWORT		ANTR1600
	TZE	HH+2		ANTR1610
	CLA	HVEL		ANTR1620
	STO	VEL+1		ANTR1630
	CLA	TARG+2	HELIOCENTRIC	ANTR1640
	TSX	TAB,4	INTERPOLATION	ANTR1650
HH	PZE	**,,63		ANTR1660
	PZE	XN+3,,HWORT+1		ANTR1670
RESET	AXT	3,4	PLACE	ANTR1680
	CLA	XN+6,4	COORDINATES	ANTR1690
	FDH	EVEL	IN OLD	ANTR1700
	STQ	XN.+18,4	FORMAT	ANTR1710
	CLA	XN+9,4		ANTR1720
	FDH	EVEL		ANTR1730
	STQ	XN.+12,4		ANTR1740
	LDQ	XN+3,4		ANTR1750
	STQ	XN+6,4		ANTR1760
	FMP	MU		ANTR1770
	FSB	XN+18,4		ANTR1780
	STO	XN+9,4	POSITION OF SUN	ANTR1790
	STZ	XN+3,4		ANTR1800
	LDQ	XN.+3,4		ANTR1810
	STQ	XN.+6,4		ANTR1820
	FMP	MU		ANTR1830
	FSB	XN.+18,4		ANTR1840
	STO	XN.+9,4	VELOCITY OF SUN	ANTR1850
	STZ	XN.+3,4		ANTR1860
	CLA	SATPOS+3,4		ANTR1870
	STO	XN+18,4		ANTR1880
	CLA	SATVEL+3,4		ANTR1890
	STO	XN.+18,4		ANTR1900
	TIX	RESET+1,4,1		ANTR1910
	LXA	CENTER,4		ANTR1920
	LXD	VEL,2	IX2 NOT ZERO IF VELOCITY OPTION	ANTR1930
	TXH	RVPRT,2,0	GET COORDINATES FOR PRINTING	ANTR1940
	TRA	RVPRT		ANTR1950
	CAL	KB6		ANTR1960
	ORA	KB6-1		ANTR1970
	TZE	FLEE		ANTR1980
	AXT	3,1		ANTR1990
	CLA	XN+21,1		ANTR2000
	FAD	XN+9,1		ANTR2010
	STO	XN+21,1		ANTR2020
	CLA	XN+18,1		ANTR2030
	FAD	XN+9,1		ANTR2040
	STO	XN+18,1		ANTR2050
	TIX	*-6,1,1		ANTR2060
	TRA	FLEE		ANTR2070
TESTM	TXH	RVPRT,4,1		ANTR2080
	AXT	3,1	MOON-CENTERED	ANTR2090
	CLS	XN+6,1		ANTR2100
	STO	XN+3,1		ANTR2110
	CLA	XN+9,1		ANTR2120
	FSB	XN+6,1		ANTR2130
	STO	XN+9,1		ANTR2140
	STZ	XN+6,1		ANTR2150
	TIX	*-6,1,1		ANTR2160
FLEE	CLA	CENTER		ANTR2170
	STO	KERNO		ANTR2180
	CLA	TARG		ANTR2190
	STO	TARGO		ANTR2200
	CLA	TARG+1		ANTR2210
	STO	TARGO+1		ANTR2220
	AXT	21,1		ANTR2230
	CLA	XN+21,1		ANTR2240

```

OUP  STO  **,1
     TIX  OUP-1,1,1
     CLA  VEL
     TZE  OUT
     AXT  21,1
     CLA  XN.+21,1
XN.V  STO  **,1
     TIX  XN.V-1,1,1
OUT   LXN  HELIO-1,1
     LXN  TRAP+1,2
     LXN  TRAP,4
     TRA  1,4
     REM
     REM  TSX  TAB,4
     REM  PZE  B,,K
     REM  PZE  A,,C
     REM
     REM
     REM
     REM
     REM
TAB   SXD  COM+9,4
     SXD  COM+8,2
     SXD  COM+7,1
     STO  ARG
     CLA  1,4
     STA  TAB18
     LRS  18
     ADD  1,4
     STA  TAB21
     CLA  2,4
     PAX  ,1
     TXI  *+1,1,SEP
     SXA  TAB29,1
     ARS  18
     STA  VELOP
     AXT  2,1
     NZT  VEL
     TXI  VELOP,1,-1
VELOP CLA  **,1
     STO  VEL+2
     STZ  SATURN
     ANA  =03
     SUB  =02
     TNZ  *+3
     CLA  =2
     STO  SATURN
     TXL  POSOP,1,1
     REM
     CLA  ARG
     TSX  COEFF,,4
     AXT  3,4
     CLS  COM+13,4
     STO  COM+16,4
     TIX  *-2,4,1
     CLA  1.
     FSB  ARG
     TSX  COEFF,,4
     TRA  TAB1-1
     REM
POSOP CLA  ARG
     TSX  COEFF,4
     AXT  3,4
     CLA  COM+13,4
     STO  COM+16,4
     TIX  *-2,4,1
     CLA  1.
     FSB  ARG
     TSX  COEFF,4
     LXN  COM+9,4
     CLA  2,4
     STA  TAB29
     SXD  COM+4,1
TAB1  AXT  0,2
     AXT  0,1

```

(AC)=INTERPOLATIVE ARGUMENT

B=START OF DATA BLOCK
K=WORDS PER SUB BLOCK
A=START OF RESULT BLOCK
C=SKIP CODE WORD LOCATION

PICK UP SKIP
CODE WORD

TEST FOR INCLUSION OF SATURN
WHENEVER JUPITER IS PRESENT

VELOCITY OPTION

FORM THE
E1(2J).

FORM THE
E0(2J).

POSITION OPTION

FORM THE
E1(2J)

FORM THE
E0(2J)

REFER TO
POSITION
STORAGE

```

ANTR2250
ANTR2260
ANTR2270
ANTR2280
ANTR2290
ANTR2300
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ANTR2980
ANTR2990

```

	LDQ	VEL+2		ANTR3000
	ZAC			ANTR3010
	LGL	2		ANTR3020
	PAX	4		ANTR3030
	STQ	VEL+2		ANTR3040
	CAL	=03		ANTR3050
	ORS	VEL+2		ANTR3060
	TXH	END,4,2		ANTR3070
	TXH	FORT,4,0		ANTR3080
	TXI	*+1,1,-3		ANTR3090
	TXI	TAB1+2,2,-1		ANTR3100
FORT	AXT	3,4		ANTR3110
TAB18	LDQ	**1	X(0)	ANTR3120
	FMP	COM+13,4		ANTR3130
	STO	COM		ANTR3140
TAB21	LDQ	**1	X(1)	ANTR3150
	FMP	COM+16,4		ANTR3160
	FAD	COM		ANTR3170
	STO	COM+4,4		ANTR3180
	TXI	*+1,1,-1		ANTR3190
	TIX	TAB18,4,1		ANTR3200
	FAD	COM+2		ANTR3210
	FAD	COM+1		ANTR3220
TAB29	STO	**2	X(T)	ANTR3230
	TXI	TAB1+2,2,-1		ANTR3240
END	NZT	SATURN		ANTR3250
	TRA	END1		ANTR3260
	TXH	END1,2,-4	COMPLETED SATURN	ANTR3270
	LXD	COM+4,4		ANTR3280
	CLA	TABSAT+2,4		ANTR3290
	STA	TAB29		ANTR3300
	AXT	0,2		ANTR3310
	CAL	=0537000000000	TWO ADDITIONAL COORDINATES	ANTR3320
	SLW	VEL+2		ANTR3330
	TRA	FORT		ANTR3340
END1	LXD	COM+4,1		ANTR3350
	TIX	VELOP,1,1		ANTR3360
	LXD	COM+9,4		ANTR3370
	LXD	COM+8,2		ANTR3380
	LXD	COM+7,1		ANTR3390
	TRA	3,4		ANTR3400
COEFF	STO	COM+10	CALCULATE	ANTR3410
	LDQ	COM+10	POSITION	ANTR3420
	FMP	COM+10	COEFFICIENTS	ANTR3430
	STO	COM+12	FOR EVERETT,S	ANTR3440
	FSB	1.	INTERPOLATION	ANTR3450
	FDH	6.	FORMULA	ANTR3460
	FMP	COM+10		ANTR3470
	STO	COM+11		ANTR3480
	CLA	COM+12		ANTR3490
	FSB	4.		ANTR3500
	FDH	20.		ANTR3510
	FMP	COM+11		ANTR3520
	STO	COM+12		ANTR3530
	TRA	1,4		ANTR3540
COEFF.	STO	COM	CALCULATE	ANTR3550
	CLS	1.	VELOCITY	ANTR3560
	FDH	VEL+1	COEFFICIENTS	ANTR3570
	STQ	COM+10	FOR EVERETT,S	ANTR3580
	LDQ	COM	INTERPOLATION	ANTR3590
	FMP	COM	FORMULA	ANTR3600
	STO	COM+12		ANTR3610
	XCA			ANTR3620
	FMP	3.		ANTR3630
	FSB	1.		ANTR3640
	FDH	6.		ANTR3650
	FMP	COM+10		ANTR3660
	STO	COM+11		ANTR3670
	CLA	COM+12		ANTR3680
	FSB	3.		ANTR3690
	XCA			ANTR3700
	FMP	5.		ANTR3710
	XCA			ANTR3720
	FMP	COM+12		ANTR3730
	FAD	4.		ANTR3740

	FDH 120.	
	FMP COM+10	
	STO COM+12	
	TRA 1,4	
RVPRT	AXT 0,2	EXPRESS ALL
	AXT 3,4	BODIES GEOCENTRICALLY
	AXT 3,1	
	CLA XN+9,2	
	FAD XN+9,4	
	STO XN+9,2	
	CLA VEL	
	TZE **4	
	CLA XN.+9,2	
	FAD XN.+9,4	
	STO XN.+9,2	
	TXI **1,2,-1	
	TIX RVPRT+3,4,1	
	TXH RVPRT+1,2,9-SEP	
	CLA CENTER	BUFFER
	ALS 1	CENTRAL
	ADD CENTER	BODY
	PAC ,4	
	CLA XN,4	
	STO COM+3,1	
	CLA XN.,4	
	STO COM+6,1	
	TXI **1,4,-1	
	TIX *-5,1,1	
RVPRT1	AXT 0,2	EXPRESS ALL
	AXT 3,4	BODIES IN TERMS
	CLA XN,2	OF THE CENTRAL
	FSB COM+3,4	BODY
	STO XN,2	
	CLA VEL	
	TZE **4	
	CLA XN.,2	
	FSB COM+6,4	
	STO XN.,2	
	TXI **1,2,-1	
	TIX RVPRT1+2,4,1	
	TXH RVPRT1+1,2,-SEP	
	TRA FLEE	
LOOK	CLA TLAST	
	FSB TARG	
	TNZ **2	
	TSX ERP,4	ARGUMENT TOO LARGE
	TMI *-1	
	CLA TARG	
	FSB TFIRST	
	TPL **2	
	TSX ERP,4	ARGUMENT TOO SMALL
	FDH 20.	
	STQ TEM	
	*IFIX TEM	HELIO SPACING
	STO TEM	
	*FLOAT TEM	
	XCA	
	FMP 20.	
	FAD TFIRST	
	STO COM+1	TIME ON RECORD
	CLA TABLE	
	FSB COM+1	
	TMI FINDIT	
	FDH 20.	
	STQ TEM	
	*IFIX TEM	
	ADD 1F	RECORDS TO
	PAX ,1	BE BACKED
	TXL BRS,1,20	OVER
	TSX .REW,4	
FILE	PZE 0	
	TRA FINDIT	
BRS	TSX .BSR,4	
FLE	PZE 0,0,ERR	
	TIX BRS,1,1	

ANTR3750
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 ANTR3990
 ANTR4000
 ANTR4010
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 ANTR4090
 ANTR4100
 ANTR4110
 ANTR4120
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 ANTR4240
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 ANTR4470
 ANTR4480
 ANTR4490

FINDIT	AXT	10,2			ANTR4500
	TSX	.READ,4			ANTR4510
FEL	PZE	*			ANTR4520
	PZE	ERR,0,ERR			ANTR4530
	IOCT	TABLE,,569			ANTR4540
	CLA	TABLE			ANTR4550
	SUB	COM+1			ANTR4560
	TMI	*+2			ANTR4570
	TNZ	LOOK	TAPE NOT POSITIONED PROPERLY		ANTR4580
	TNZ	FINDIT			ANTR4590
	AXT	567,4			ANTR4600
	CAL	TABLE	CHECK		ANTR4610
	ACL	A+567,4	SUM		ANTR4620
	TIX	*-1,4,1			ANTR4630
	LAS	C			ANTR4640
	TRA	*+2			ANTR4650
	TRA	SCALE			ANTR4660
BSRA	TSX	.BSR,4			ANTR4670
FLEA	PZE	0,0,ERR			ANTR4680
	TIX	FINDIT+1,2,1			ANTR4690
ERR	CALL	.FWRD.(FLUT,FMT)			ANTR4700
ERRR	TSX	.FFIL.,4			ANTR4710
	CALL	EXIT			ANTR4720
ERP	CALL	.FWRD.(FLUT,FMT1)			ANTR4730
	CLA	TARG			ANTR4740
	TSX	.FCNV.,4			ANTR4750
	TRA	ERRR			ANTR4760
FMT	BCI	7,(33HOTAPE ERROR IN ANTR. EXIT CALLED.)			ANTR4770
FLUT	PZE	0	SYSTEM OUTPUT FILE ADDRESS		ANTR4780
FMT1	BCI	7,(7H0TIME =,E20.8,21H OUT OF RANGE IN ANTR)			ANTR4790
SCALE	AXT	189,4			ANTR4800
	LDQ	A+189,4	SCALE		ANTR4810
	FMP	SCALE1	GEOCENTRIC		ANTR4820
	STO	A+189,4	EPHEMERIS		ANTR4830
	TIX	*-3,4,1			ANTR4840
	AXT	378,4			ANTR4850
	LDQ	B+378,4	SCALE		ANTR4860
	FMP	SCALE2	HELIOCENTRIC		ANTR4870
	STO	B+378,4	EPHEMERIS		ANTR4880
	TIX	*-3,4,1			ANTR4890
	CLA	EGM			ANTR4900
	FAD	MGM			ANTR4910
	STO	COM	OF MOON		ANTR4920
	CLA	MGM			ANTR4930
	FDH	COM	BARYCENTER		ANTR4940
	STQ	MU	FOR POSITION OF EARTH		ANTR4950
	TRA	NEU			ANTR4960
SCALE1	DEC	6378.165	EARTH RADIUS		ANTR4970
SCALE2	DEC	149599000.	ASTRONOMICAL UNIT		ANTR4980
GRAV	DEC	3.986032E5	EARTH		ANTR4990
	DEC	4.900759E3	MOON		ANTR5000
	DEC	1.32715445E11	SUN		ANTR5010
	DEC	3.247695E5	VENUS		ANTR5020
	DEC	4.297780E4	MARS		ANTR5030
	DEC	3.791870E7	SATURN		ANTR5040
	DEC	1.267106E8	JUPITER		ANTR5050
TEMPDT	DEC	34.,0	E.T.-U.T.		ANTR5060
TFIRST	DEC	10.0	0 HR JAN 11,1950 JD=2433292.5		ANTR5070
TLAST	DEC	11530.0	0 HR JUL 27,1981 JD=2444812.5		ANTR5080
	RJ	DEC 1E6	JUPITER TEST DISTANCE		ANTR5090
EVEL	DEC	86400.			ANTR5100
HVEL	DEC	345600.			ANTR5110
MU	PZE		MASS RATIO OF MOON TO BARYCENTER		ANTR5120
	OCT	000000520052	MARS, JUPITER VELOCITY		ANTR5130
HWORT	OCT	0			ANTR5140
	OCT	527777777777	MOON VELOCITY		ANTR5150
EWORT	OCT	0			ANTR5160
HWORTJ	OCT	000000005252	BARYCENTER, JUPITER		ANTR5170
HWORTE	OCT	000000005200	BARYCENTER		ANTR5180
EWORTO	OCT	527777777777			ANTR5190
HWORTO	OCT	000052525252			ANTR5200
HWORTV	OCT	525252525252			ANTR5210
TARGO	DEC	20.,0	FORMER TIME		ANTR5220
KERNO	PZE		FORMER CENTER		ANTR5230
	PZE	A	GEOCENTRIC REFERENCE		ANTR5240

	PZE	B
ARG		
SATURN	DEC	0
TABSAT	PZE	SATVEL
	PZE	SATPOS
1F	DEC	1
1.	DEC	1.
3.	DEC	3.
4.	DEC	4.
5.	DEC	5.
6.	DEC	6.
20.	DEC	20.
120.	DEC	120.
86400.	DEC	86400.
VEL	BSS	3
SATPOS	BSS	3
SATVEL	BSS	3
TRAP	BSS	2
*		
TABLE	DEC	0
A	BSS	189
B	BSS	378
C	BSS	1
*		
EPHTAB	SYN	TABLE+568
COM	BSS	21
TARG	PZE	
	PZE	
	PZE	
CENTER	PZE	
KBO	PZE	
KB2	PZE	
KB6	PZE	
R	PZE	
XN	BSS	21
XN.	BSS	21
TEM	PZE	
LDIR	LDIR	
TAPNO	PZE	8
T	PZE	
SEP	SYN	XN.-XN
BSEP	SYN	7
END		

HELIOCENTRIC REFERENCE

0=NOT NEEDED,OTHERWISE COMPUTE SATURN

VELOCITY OPTION,H,SKIP CODE

RESERVE
FOR
WORKING
EPHEMERIS

ANTR5250
ANTR5260
ANTR5270
ANTR5280
ANTR5290
ANTR5300
ANTR5310
ANTR5320
ANTR5330
ANTR5340
ANTR5350
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ANTR5370
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ANTR5630
ANTR5640
ANTR5650
ANTR5660
ANTR5670

Subroutine: ARKTNS

Purpose: To obtain the arctangent of Y/X. The solution is obtained between ± 180 degrees if N is set equal to 180. If $N \neq 180$, the angle is computed between 0 and 360.

Calling Sequence: Z = ARKTNS(N,X,Y)

Input and Output

I/ ϕ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	N				Range key (see above).
I	X				
I	Y				
ϕ	Z			radians	$\tan^{-1}(Y/X)$

Common storages used: None

Subroutines required: None

ARKTNS-1

\$IBFTC MC13AT XR3,M94,NOREF,NODD	
FUNCTION ARKTNS(N,X,Y)	ATNS0001
C COMPUTES 4-QUADRANT ARCTANGENT OF Y/X IN RADIANS	ATNS0002
C N=360 ANGLE LIES IN RANGE (0,360) DEG	ATNS0003
C N=180 ANGLE LIES IN RANGE (-180,180) DEG	ATNS0004
C USES ATAN2, ALLOWS 0/0=0	ATNS0005
IF(X)100,10,100	ATNS0006
10 IF(Y) 20,30,40	ATNS0007
20 ARKTNS =-1.5707963	ATNS0008
GO TO 200	ATNS0009
30 ARKTNS= 0.0	ATNS0010
35 RETURN	ATNS0011
40 ARKTNS=1.5707963	ATNS0012
GO TO 35	ATNS0013
100 ARKTNS = ATAN2(Y,X)	ATNS0014
200 IF(N-180)205,35,205	ATNS0015
205 IF(ARKTNS)210,35,35	ATNS0016
210 ARKTNS=ARKTNS+6.28318531	ATNS0017
GO TO 35	ATNS0018
END	

Subroutine: **BACK**

Purpose: **To backspace a binary tape N logical records or a
BCD tape N physical records.**

Calling Sequence: **CALL BACK(M,N)**

Input and Output

I/φ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	M				Tape logical number.
I	N				Number of records to be backspaced.

Common storages used: None

Subroutines required: None

BACK-1

```
$IBFTC MC13J3 XR3,M94,NODD
  SUBROUTINE BACK(M,N)
C THIS SUBROUTINE BACKSPACES BINARY TAPE M,N LOGICAL RECORDS
C OR BCD TAPE M, N PHYSICAL RECORDS
  DO 100 I=1,N
    BACKSPACE M
100 CONTINUE
  RETURN
  END
```

```
BACK0001
BACK0002
BACK0003
BACK0004
BACK0005
BACK0006
BACK0007
BACK0008
```

Subroutine: BARN

Purpose: Random number generator. Provides either uniform random variable over the interval (0, 1) or a normal random number with zero mean and unit variance.

Calling Sequence: Y = BARN (I)

Input and Output

I/ ϕ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	I				+1, uniform distribution -1, normal distribution
ϕ	Y				Random number

Common storages used: None

Subroutines required: None

BARN-1

\$IBMAP	MC13B4		
BARN	SAVE	(1,2)	BARN0000
	CLA	1,4	BARN0001
	PDX	0,2	BARN0002
	TXL	GG,2,1	BARN0003
	CLA*	4,4	BARN0004
	TMI	GG	BARN0005
	CLA*	5,4	BARN0006
	STO	HERE	BARN0007
GG	CLA*	3,4	BARN0008
	TMI	RDNN	BARN0009
RDUN	LDQ	HERE	BARN0010
	MPY	HERE+1	BARN0011
	LLS	4,0	BARN0012
	ALS	4,0	BARN0013
	LRS	4,0	BARN0014
	STQ	HERE	BARN0015
	ADD	HERE	BARN0016
	STO	HERE	BARN0017
	ARS	4,0	BARN0018
	ORA	HERE+2	BARN0019
	FAD	HERE+2	BARN0020
SWITCH	NOP	NEXT	BARN0021
	TRA	RETURN	BARN0022
HERE	OCT	2312421637,1737,200000000000	BARN0023
RDNN	LXA	L20,1	BARN0024
	CLA	HERE+2	BARN0025
	STO	C,0	BARN0026
	LDQ	TRA	BARN0027
	SLQ	SWITCH	BARN0028
	TRA	RDUN	BARN0029
NEXT	LDQ	NOP	BARN0030
	SLQ	SWITCH	BARN0031
	FAD	C,0	BARN0032
	STO	C,0	BARN0033
	TIX	NEXT-3,1,1	BARN0034
	FDP	L20+1,0	BARN0035
	CLA	L20+4,0	BARN0036
	LLS	35,0	BARN0037
	FSB	L20+2,0	BARN0038
	LRS	35,0	BARN0039
	FMP	L20+3,0	BARN0040
RETURN	TXL	RETRN,2,1	BARN0041
	LDQ	HERE	BARN0042
	STQ*	5,4	BARN0043
RETRN	RETURN	BARN	BARN0044
L20	HTR	20,0	BARN0045
	DEC	20,,.5,15.49193340,0	BARN0046
TRA	OCT	+002000000000	BARN0047
NOP	OCT	+076100000000	BARN0048
C	HTR	0,0	BARN0049
	END		BARN0050

Subroutine: BECSTA

Purpose: Computes start and stop time arrays for stations and beacons.

Calling Sequence: CALL BECSTA(KSTA,KBEA,KMAXS,KMAXB)

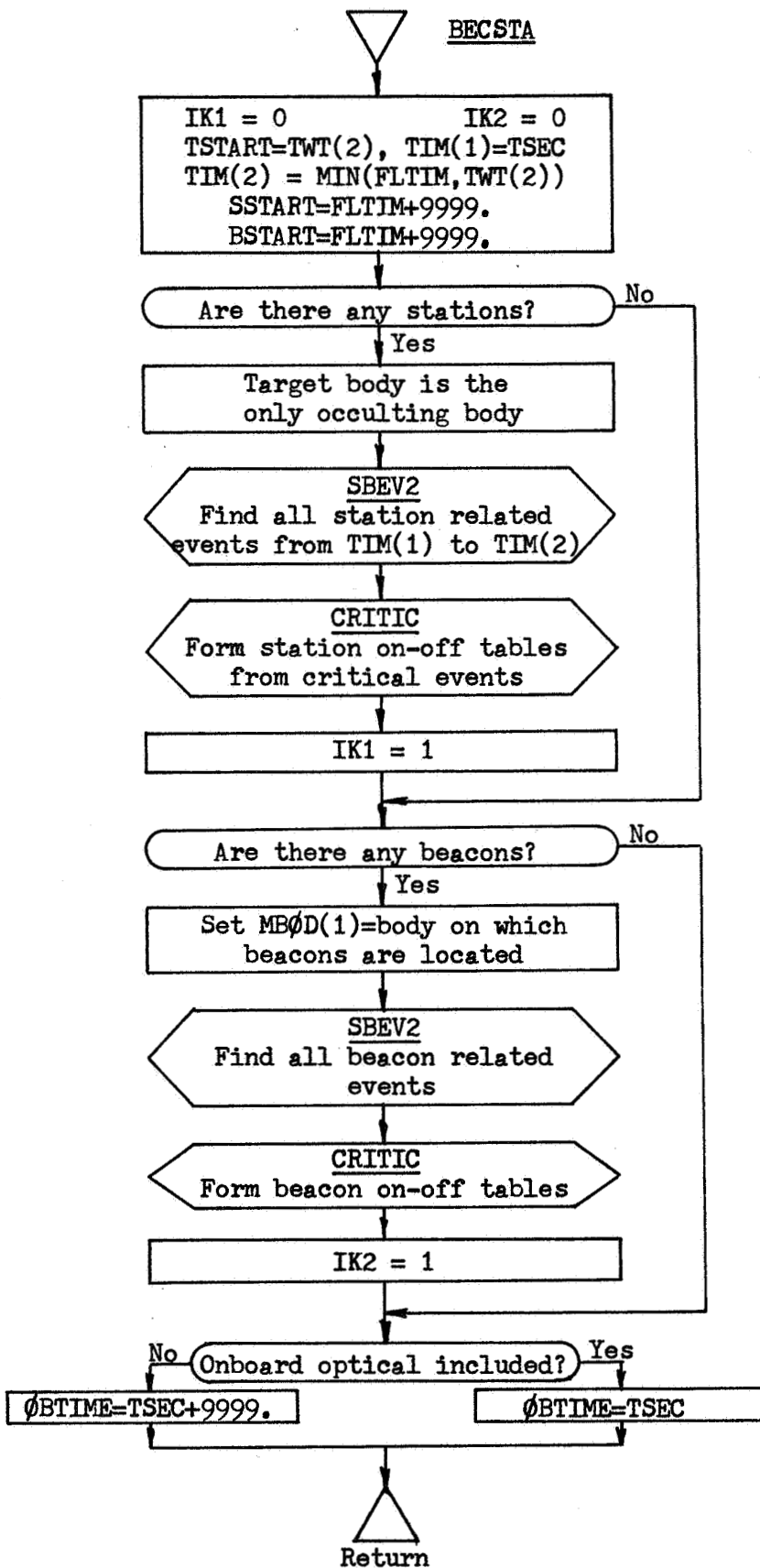
Input and Output

I/φ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
φ	KSTA	75			Array of station numbers corresponding to times in SSTART,SSTOP arrays.
φ	KBEA	75			Array of beacon numbers corresponding to times in BSTART,BSTOP arrays.
φ	KMAXS				Number of station on-off time pairs found.
φ	KMAXB				Number of beacon on-off time pairs found.

Common storages used: /INPCOM/,/WCOM/,/EXIC/

Subroutines required: CRITIC,EXINST,SBEV2

BECSTA-1



BECSTA-2

```

$IBFTC MC13BA NOREF,M94,NODD,XR3
CMC13BA BECSTA COMPUTES START AND STOP ARRAYS FOR STATIONS AND BEACONS
SUBROUTINE BECSTA(KSTA,KBEA,KMAXS,KMAXB)
C SUBRS REQUIRED SBEV2
COMMON/INPCOM/C(700)/WCOM/IW(550),CW(1450)
COMMON/EXIC/W(36),CRX(9,2)
DIMENSION TWT(3), TIM(2), STIME(50), ISC(12)
1, MBOD(4), XIN(6), TI(3), S(23,12)
2, SECR(12), ISEE(12), ISTIM(50), BTIME(50)
3, IBC(10), IB(36), B(91), BECR(10)
4, IBEE(10), IBTIM(50), IOBA(3,6)
DIMENSION SSTART(75), SSTOP(75), KSTA(75)
1, BSTART(75), BSTOP(75), KBEA(75)
EQUIVALENCE (C(200),S), (C(476),B)
EQUIVALENCE (IW(8),ISC), (IW(20),IBC), (IW(58),IOBA)
1, (IW(312),IB), (IW(355),ISEE), (IW(367),IBEE)
2, (IW(382),NOR), (IW(383),ITARG)
3, (IW(384),IK1), (IW(385),KSMAX), (IW(386),IK2)
4, (IW(387),KBMAX), (IW(389),ISTIM), (IW(439),IBTIM)
5, (IW(489),ITRIG), (IW(490),KOUNT)
EQUIVALENCE (CW(1134),XIN), (CW(1140),TSEC)
1, (CW(1160),TI), (CW(1175),OBTIME), (CW(1178),TSTART)
2, (CW(1179),FLTIM), (CW(1188),TWT), (CW(1441),BECR)
3, (CW(1200),STIME), (CW(1250),BTIME), (CW(1360),SECR)
4, (CW(222),SSTART), (CW(297),SSTOP)
5, (CW(372),BSTART), (CW(447),BSTOP)
DOUBLE PRECISION TWR
IK1=0
IK2=0
TSTART = TWT(2)
TIM(1) = TSEC
TIM(2) = TWT(2)
IF(TIM(2).GT.FLTIM) TIM(2)=FLTIM
SSTART=FLTIM+9999.
IF(ISC .EQ. 0) GO TO 24
MBOD(2) = ITARG
IF(ITARG .EQ. 1) MBOD(2) = 0
MBOD(3) = 0
ITRIG=1
CALL SBEV2(XIN,TI,NOR,TIM,S,23,12,ISC,SECR,ISEE,ISTIM,STIME,MBOD)
IF(KOUNT.LE.0) GO TO 2
DO 1 I=1,KOUNT
1 BACKSPACE 10
M=0
N=18
TWR=TSEC
CALL EXINST(TWR,M,N,XIN,XIN(4),W,CRX,10,ITRIG,KOUNT)
2 KSMAX=MBOD(4)
ITRIG=0
KOUNT=0
CALL CRITIC(12,ISC,STIME,ISTIM,ISEE,KSMAX,TIM,SSTART,SSTOP,KSTA,
IKMAXS)
IK1 = 1
24 CONTINUE
BSTART=FLTIM+9999.
IF(IBC .EQ. 0) GO TO 25
MBOD(1) = IB(36)
MBOD(2) = 0
ITRIG=1
CALL SBEV2(XIN,TI,NOR,TIM,B,8,10,IBC,BECR,IBEE,IBTIM,BTIME,MBOD)
IF(KOUNT.LE.0) GO TO 241
DO 240 I=1,KOUNT
240 BACKSPACE 10
M=0
N=18
TWR=TSEC
CALL EXINST(TWR,M,N,XIN,XIN(4),W,CRX,10,ITRIG,KOUNT)
241 KBMAX=MBOD(4)
ITRIG=0
KOUNT=0
CALL CRITIC(10,IBC,BTIME,IBTIM,IBEE,KBMAX,TIM,BSTART,BSTOP,KBEA,
IKMAXB)
IK2 = 1
25 CONTINUE
OBTIME = TSEC
BCST0001
BCST0002
BCST0003
BCST0004
BCST0005
BCST0006
BCST0007
BCST0008
BCST0009
BCST0010
BCST0011
BCST0012
BCST0013
BCST0014
BCST0015
BCST0016
BCST0017
BCST0018
BCST0019
BCST0020
BCST0021
BCST0022
BCST0023
BCST0024
BCST0025
BCST0026
BCST0027
BCST0028
BCST0029
BCST0030
BCST0031
BCST0032
BCST0033
BCST0034
BCST0035
BCST0036
BCST0037
BCST0038
BCST0039
BCST0040
BCST0041
BCST0042
BCST0043
BCST0044
BCST0045
BCST0046
BCST0047
BCST0048
BCST0049
BCST0050
BCST0051
BCST0052
BCST0053
BCST0054
BCST0055
BCST0056
BCST0057
BCST0058
BCST0059
BCST0060
BCST0061
BCST0062
BCST0063
BCST0064
BCST0065
BCST0066
BCST0067
BCST0068
BCST0069
BCST0070
BCST0071
BCST0072
BCST0073
BCST0074

```

IF(IOBA .EQ. 0) OBTIME = FLTIM +9999.
RETURN
END

BCST0075
BCST0076
BCST0077

Subroutine: BIBCD

Purpose: To convert a binary integer to BCD.

Calling Sequence: BCD = BIBCD(BIN)

Input and Output

I/φ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	BIN				Full word binary integer $0 \leq \text{BIN} < 10^{-6}$
φ	BCD				BCD equivalent of BIN with leading zeros replaced by blanks.

Common storages used: None

Subroutines required: None

BIBCD-1

\$IBMAP	MC1316	50,NODD		
	TTL	16	CONVERSION OF POSITIVE FIXED-POINT NUMBERS	
RIBCD	SAVE			BIBC0001
	CLA*	3,4		BIBC0002
	TNZ	NZ		BIBC0003
	CLA	BLNKZ		BIBC0004
RETRN	RETURN	BIBCD		BIBC0005
NZ	SSP			BIBC0006
	STO	SV		BIBC0007
	AXT	6,4		BIBC0008
	CLA	TEMP		BIBC0009
	STO	TEMP+7,4		BIBC0010
	TIX	*-1,4,1		BIBC0011
	AXT	6,4		BIBC0012
	LDQ	SV		BIBC0013
NDN	PXA	0,0		BIBC0014
	DVP	TEN		BIBC0015
	STQ	SV		BIBC0016
	ALS	30		BIBC0017
	STP	TEMP+7,4		BIBC0018
	ORS	TEMP+7,4		BIBC0019
	NZT	SV		BIBC0020
	TRA	DN		BIBC0021
	TIX	NDN,4,1		BIBC0022
DN	LDQ	MQTST		BIBC0023
	CRQ	TEMP,,6		BIBC0024
	XCA			BIBC0025
	RETURN	BIBCD		BIBC0026
MQTST	OCT	060504030201		BIBC0027
SV	PZE	0		BIBC0028
RLNKZ	OCT	606060606000		BIBC0029
TEMP	SIX	*		BIBC0030
	BSS	6		BIBC0031
TEN	PZE	10		BIBC0032
	END			BIBC0033

Subroutine: BLDCØV

Purpose: To build up square submatrices of the covariance matrix from standard deviations and normalized correlations and to build subsets of the vector of state variable BCD names from an array of possible names.

Calling Sequence: CALL BLDCØV(S,TITLE,KHØLD,NUMBER,LØCAT)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	S	d			Vector of standard deviations and correlations.
I	TITLE				Vector of possible state names.
I	KHØLD				Construction of P will begin with element P(KHØLD+1,KHØLD+1).
I	NUMBER				Dimension of submatrix P to be built on this entry.
I	LØCAT				Array associated with S, used for choosing the desired element of S.

Common Storages Used: /ES1CØM/
Subroutines Required: None.

BLDCØV-1

1. /ESLCOM/. The arrays of ESLCOM (q.v.) used by BLDCOV are

ϕ PNEW d(N,N) Covariance matrix, P
 ϕ STNAMN (N) Vector of state names
I ILCN (N) Vector of state element identification
indices

where N is the maximum length of the state vector.

2. Usage

The allowed state variables for the Differential Correction Program are divided into logically related groups. BLDCOV is called for each of the variable groups to be included in the state, and for which the related portion of the covariance matrix is to be built up from block data arrays of standard deviations and normalized correlations. BLDCOV is used only at case initialization and is called only by SETCAS. The resulting initial covariance matrix for a case allows correlations between variables within the same variable group; correlations between variables in different groups are set to zero by SETCAS. For example, if the state consisted of variables from each of three groups, the built-up initial covariance matrix would have the following form:

P_1	ϕ	ϕ
ϕ	P_2	ϕ
ϕ	ϕ	P_3

where ϕ represents the null matrix. Each of the blocks P_i , $i = 1, 3$ is built separately. Each requires a call of BLDCOV if it is to be built from stored block data. (Such a block might, instead, be lifted intact from a previous tape-stored estimate. This choice is exercised in SETCAS.) The variable groups and the related arrays used in the calls of BLDCOV are as follows:

BLDCOV-2

Group Code Name	Description	S Vector of Standard Deviations and Correlations	TITLE Vector of BCD Element Names	LØCAT Vector of Indices for Using S
SPCDAT	Spacecraft position and velocity	SPCDAT	SPCNAM	LØCSPC
EFEMRS	Planetary mu's and ephemeris parameters	EFEDAT	EFENAM	LØCEFE
PRESUR	Solar pressure, drag, venting	PREDAT	PRENAM	LØCPRE
DELTIM	Ephemeris to universal time conversion	DELDAT	DELNAM	LØCDEL
HARMO3	Gravitational harmonics for Earth	EHADAT	HARNAM	LØCHAR
HARM11	Moon harmonics	MHADAT	HARNAM	LØCHAR
HARM ¹ _{mn}	Extra body harmonics	XHADAT	HARNAM	LØCHAR
(Name) ²	Tracking station biases	SEN(1,K) ³	STBNAM	LØCSTA

¹Where mn is a body number other than 03 or 11.

²Any valid station name.

³Where K is the internal number assigned to the station by subroutine SETSTA.

BLDCØV-3

At each call BLDCOV will construct the square portion beginning in P(KHOLD+1,KHOLD+1) and ending in P(KHOLD+NUMBER,KHOLD+NUMBER). BLDCOV will construct the subset from STNAMN(KHOLD+1) to STNAMN(KHOLD+NUMBER). At any call of BLDCOV, KHOLD is the size of the covariance matrix already built, NUMBER is the number of additional rows and columns to be added by this call.

The use of the arrays S, TITLE, ILOCN, and LOCAT is illustrated by the following example:

Suppose that the first six rows and columns of P have already been built. Then KHOLD = 6. The first elements of ILOCN contain the identifying numbers for the six state variables for which P has already been built. Now, suppose that the next group to be added is the solar pressure, drag, vent group. Let us assume that the two drag parameters are to be included. Then NUMBER = 2 and ILOCN(7) = 2, ILOCN(8) = 3. The various arrays are as follows:

```

ILOCN = (k1, k2, k3, k4, k5, k6, 2, 3, n1, m2, . . . )
TITLE = (3HSPR, 3HDR1, 3HDR2, 3HVNT) = PRENAM
KHOLD = 6
NUMBER = 2
LOCAT = (7, 9, 10) = LOCPRE
S = PREDAT:
  S(1) = Solar Pressure value
  (2) = Drag 1           "
  (3) = Drag 2           "
  (4) = Venting          "
  (5) = Solar Pressure Standard Deviation
  (6) = Drag 1           "      "
  (7) = Drag 2           "      "
  (8) = Venting          "      "
  (9) = C(1,2) , correlation between Solar Pressure and Drag 1
  (10) = C(1,3)
  (11) = C(1,4)
  (12) = C(2,3)
  (13) = C(2,4)
  (14) = C(3,4)

```

BLDCOV-4

The variable identifying numbers to be used are $ILOCN(7) = 2$ and $ILOCN(8) = 3$.

BLDCOV first picks up elements 2 and 3 from the TITLE array and puts them in positions 7 and 8 of the STNAMN array.

Next, the diagonal terms $P(7,7)$ and $P(8,8)$ of the covariance matrix are loaded with the correct standard deviations from the S array. For this purpose, the LOCAT array is used. Let K be the integer part of $LOCAT(1)/2 + 1$. That is, $K = 4$. The values 2 and 3 from the ILOCN array are added to K to yield 6 and 7. Then $P(7,7)$ and $P(8,8)$ are loaded with $S(6)$ and $S(7)$, respectively. For $P(7,8)$ we wish to load $C(2,3)$. We use the first index of C, viz., 2, to find $LOCAT(2) = 9$. To this 9 we add the second index of C, viz., 3. This yields 12. Then $S(12)$ is loaded into $P(7,8)$. Also $P(8,7)$ is set to $P(7,8)$. Thus the values of the LOCAT array are seen to describe the location of the various variables in the S array.

Finally the subset of P is un-normalized by multiplying the off-diagonal terms by their related standard deviations, after which all diagonal terms are squared to form variances.

\$IBFTC MC13SX XR3,M94,NODD,LIST	
SUBROUTINE BLDCOV (S,TITLE,KHOLD,NUMBER,LOCAT)	BLDC0001
C BUILDS PORTIONS OF COVARIANCE MATRIX	BLDC0002
DOUBLE PRECISION S(1)	BLDC0003
DIMENSION LOCAT(1) ,TITLE(1)	BLDC0004
COMMON /ESI COM/PNEW(30,30),STNAMN(30),TRAKER(30)	BLDC0005
1 ,ITRETN (30),ILOCN (30),KLOCN (54)	BLDC0006
2 ,NSN (12,20)	BLDC0007
DOUBLE PRECISION PNEW	BLDC0008
C	BLDC0009
1 DO 3 I=1,NUMBER	BLDC0010
IO = KHOLD+I	BLDC0011
KK = ILOCN(IO)+LOCAT(1)/2+1	BLDC0012
PNEW(IO,IO) = S(KK)	BLDC0013
KK = ILOCN(IO)	BLDC0014
STNAMN(IO) = TITLE(KK)	BLDC0015
IF (I.GE.NUMBER) GO TO 3	BLDC0016
LL = LOCAT(KK)	BLDC0017
II = I+1	BLDC0018
DO 2 J=II,NUMBER	BLDC0019
JO = KHOLD+J	BLDC0020
MM = ILOCN(JO)+LL	BLDC0021
PNEW(IO,JO) = S(MM)	BLDC0022
2 PNEW(JO,IO) = PNEW(IO,JO)	BLDC0023
3 CONTINUE	BLDC0024
DO 5 I=1,NUMBER	BLDC0025
IO = KHOLD+I	BLDC0026
DO 4 J=1,NUMBER	BLDC0027
JO = KHOLD+J	BLDC0028
IF (I.EQ.J) GO TO 4	BLDC0029
PNEW(IO,JO) = PNEW(IO,JO)*PNEW(IO,IO)	BLDC0030
PNEW(JO,IO) = PNEW(IO,JO)	BLDC0031
4 CONTINUE	BLDC0032
5 PNEW(IO,IO) = PNEW(IO,IO)**2	BLDC0033
999 RETURN	BLDC0034
END	

Subroutine: BMEAS

Purpose: Makes a set of beacon measurements RMEAS(2-5)
at time RMEAS(1).

Calling Sequence: CALL BMEAS(NBEAC, BRDB, XOUT, RMEAS)

Input and Output

I/φ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	NBEAC				Beacon identifying number
I	BRDB	(3)			Semi-major axis, semi-minor axis, spin rate for body on which beacons are located.
I	XOUT	(6)			Spacecraft position, velocity with respect to body on which beacons are located.
φ	RMEAS	(5)			1. Time 2. Range 3. Range rate 4. Angle 1 5. Angle 2

Common storages used: /INPCOM/, /WCOM/

Subroutines required: ARKINS, BARN, CROSS, DOT, MNA, MTRN, NUTAIT,
STAT, VNORM, VTRN

BMEAS-1

```

$IBFTC MC13MB NOREF,M94,NODD,XR3
CMC13MB BMEAS LIKE SMEAS BUT FOR BEACON
SUBROUTINE BMEAS(NBEAC,BRDB,XOUT,RMEAS)
COMMON/INPCOM/C(700)/WCOM/IW(550),CW(1450)
DIMENSION OMG(3), RMEAS(5), BRDB(3), B(91), TM(9)
1, RT(3), SC(5), XOUT(6), DUM(3), AN(3,3)
2, TI(2), EN(3,3), A(3,3), X(6), XS(6)
3, US(3), XI(3), XJ(3), XK(3), R(4), IB(36)
EQUIVALENCE (C(9),SPDLT), (C(608),EMP24), (C(476),B)
1, (CW(1140),TSEC),(CW(1160),TI), (CW(1163),AN)
2, (CW(1172),TSEC),(CW(1173),GHAR)
3, (IW(53),IEMP24),(IW(312),IB), (IW(347),NB)
EQUIVALENCE (R(1),RAT), (R(2),RDOT), (R(3),ANG1), (R(4),ANG2)
1, (TM(1),XI), (TM(4),XJ), (TM(7),XK), (SC(1),US)
C SUBROUTINES REQUIRED
C ARKTNS
C BARN
C CROSS
C DOT,VNORM
C MNA
C MTRN,VTRN
C NUTAIT
C STAT
DATA (OMG(I),I=1,2)/2*0./
RMEAS(1)=TSEC
OMG(3)=BRDB(3)
GHAB=GHAR+BRDB(3)*(TSEC-TSECO)
IF(NB=2) 1,2,3
1 CALL VTRN(AN,XOUT,X)
CALL VTRN(AN,XOUT(4),X(4))
GO TO 5
2 GHAB=0.
TIME=TI(1)+TI(2)
CALL NUTAIT(TIME,WM,CR,DA,EN,EPSIL)
CALL MNA( TIME,WM,CR,DA,EPSIL,RR,G,GP,WW,EN)
CALL MTRN(EN,AN,A)
CALL VTRN(AN,XOUT,X)
CALL VTRN(AN,XOUT(4),X(4))
GO TO 5
3 DO 4 I=1,6
4 X(I)=XOUT(I)
5 CONTINUE
KBEAC=8*NBEAC-6
CALL STAT(B(KBEAC),GHAB,TM,RT,SC,BRDB)
CALL CROSS(OMG,RT,DUM)
DO 6 I=1,3
J=I+3
XS(I)=RT(I)-X(I)
6 XS(J)=DUM(I)-X(J)
RAT=VNORM(XS,US)
RDOT=DOT(US,XS(4))
DEN=VNORM(X,US)
DO 7 I=1,3
7 XK(I)=-US(I)
CALL CROSS(XK,X(4),XJ)
DEN=VNORM(XJ,XJ)
CALL CROSS(XJ,XK,XI)
ANG1=ARKTNS(360,DOT(XS,XI),DOT(XS,XJ))
CALL CROSS(XS,XK,DUM)
DEN=VNORM(DUM,DUM)
ANG2=ATAN(DOT(XS,XK)/DEN)
IF(IEMP24.NE.0) R(1)=R(1)+EMP24*R(1)/SPDLT
DO 9 I=1,4
J=I+30
IF(IB(J).EQ.0) GO TO 9
IF(IB(J).EQ.2) GO TO 8
L=86+I
R(I)=R(I)+B(L)
8 K=82+I
M=I+1
RMEAS(M)=R(I)+B(K)*BARN(-1)
9 CONTINUE
IF(IB(35).NE.0) RMEAS(1)=RMEAS(1)+B(91)
RETURN
END
BMES0001
BMES0002
BMES0003
BMES0004
BMES0005
BMES0006
BMES0007
BMES0008
BMES0009
BMES0010
BMES0011
BMES0012
BMES0013
BMES0014
BMES0015
BMES0016
BMES0017
BMES0018
BMES0019
BMES0020
BMES0021
BMES0022
BMES0023
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BMES0052
BMES0053
BMES0054
BMES0055
BMES0056
BMES0057
BMES0058
BMES0059
BMES0060
BMES0061
BMES0062
BMES0063
BMES0064
BMES0065
BMES0066
BMES0067
BMES0068
BMES0069
BMES0070
BMES0071
BMES0072
BMES0073
BMES0074

```

Subroutine: **CARDIN**

Purpose: **To read change cards and following data cards.**

Calling Sequence: **CALL CARDIN(INITS)**

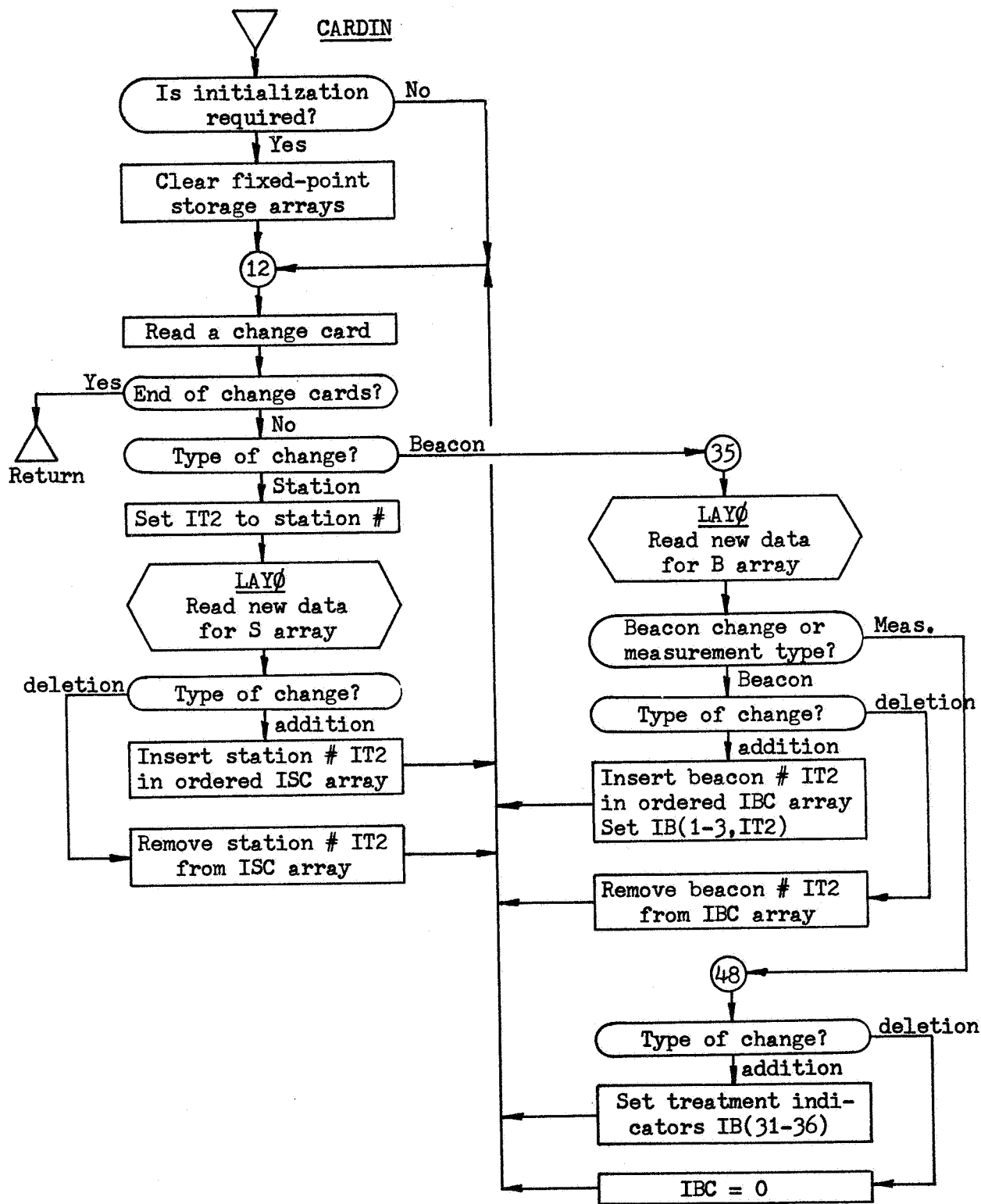
Input and Output

I/φ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	INITS				Initialization key; initializes if INITS = -1.

Common storages used: **/INPCOM/,/WCOM/**

Subroutines required: **LAYφ**

CARDIN-1



CARDIN-2

\$IBFTC MC13AR	NOREF,M94,NODD,XR3	
CMC13AR	CARDIN SAME AS PUTIN BUT CALLS LAYO	CRDN0001
	SUBROUTINE CARDIN(INITS)	CRDN0002
	COMMON/INPCOM/C(700)/WCOM/IW(550),CW(1450)	CRDN0003
	DIMENSION EMP(24), S(23,12), B(91),	CRDN0004
	OB(18)	
1,	IEMP(24), ISC(12), IBC(10), IOBR(22)	CRDN0005
2,	ITEMP(25), IS(11,12), IB(36)	CRDN0006
	EQUIVALENCE (IW(8),ISC), (IW(20),IBC), (IW(30),IEMP)	CRDN0007
1,	(IW(54),IOBR), (IW(180),IS), (IW(312),IB)	CRDN0008
2,	(C(200),S), (C(476),B), (C(567),OB)	CRDN0009
3,	(C(585),EMP)	CRDN0010
C	REQUIRED SUBROUTINES	CRDN0011
C	LAYO	CRDN0012
	IF(INITS.GE.0) GO TO 12	CRDN0013
	DO 1 I=8,75	CRDN0014
1	IW(I)=0	CRDN0015
12	CONTINUE	CRDN0016
	READ (5,102) NCH,SNAME,ITEMP	CRDN0017
102	FORMAT(I1,A6,25I2)	CRDN0018
	IF(NCH) 100,100,13	CRDN0019
13	GO TO (20,35,55,70),NCH	CRDN0020
20	CONTINUE	CRDN0021
	IT2=ITEMP(2)	CRDN0022
	CALL LAYO (S(1,IT2),NCH,ITEMP)	CRDN0023
	S(1,IT2)=SNAME	CRDN0024
	IF(ITEMP) 28,28,21	CRDN0025
21	IF(ISC.EQ.IT2) GO TO 31	CRDN0026
	NIT=0	CRDN0027
	DO 27 I=1,12	CRDN0028
	IF(ISC(I)) 23,23,24	CRDN0029
23	ISC(I)=IT2	CRDN0030
	GO TO 31	CRDN0031
24	IF(IT2-ISC(I)) 26,31,27	CRDN0032
25	IST=ISC(I)	CRDN0033
	NIT=IT2	CRDN0034
	ISC(I)=IT2	CRDN0035
26	IF(NIT.NE.IT2) GO TO 25	CRDN0036
	ITS=ISC(I+1)	CRDN0037
	ISC(I+1)=IST	CRDN0038
	IF(ITS.EQ.0) GO TO 31	CRDN0039
	IST=ITS	CRDN0040
27	CONTINUE	CRDN0041
	GO TO 31	CRDN0042
28	CONTINUE	CRDN0043
	DO 29 I=1,11	CRDN0044
	IF(IT2-ISC(I)) 12,30,29	CRDN0045
29	CONTINUE	CRDN0046
	ISC(12)=0	CRDN0047
	GO TO 12	CRDN0048
30	DO 301 J=I,11	CRDN0049
301	ISC(J)=ISC(J+1)	CRDN0050
	GO TO 12	CRDN0051
31	CONTINUE	CRDN0052
	DO 32 I=1,11	CRDN0053
32	IS(I,IT2)=ITEMP(I+2)	CRDN0054
	GO TO 12	CRDN0055
35	CONTINUE	CRDN0056
	IT2=8*(ITEMP(2)-1)+1	CRDN0057
	IF(ITEMP(2).EQ.0) IT2=81	CRDN0058
	CALL LAYO (B(IT2),NCH,ITEMP)	CRDN0059
	IF(IT2.NE.81) B(IT2)=SNAME	CRDN0060
	IT2=ITEMP(2)	CRDN0061
C	ITEMP(2)=0 INDICATES A MEASUREMENT TYPE CHANGE CARD	CRDN0062
	IF(IT2.EQ.0) GO TO 48	CRDN0063
	IF(ITEMP)43,43,36	CRDN0064
36	IF(IBC.EQ.IT2) GO TO 46	CRDN0065
	NIT=0	CRDN0066
	DO 42 I=1,9	CRDN0067
	IF(IBC(I)) 38,38,39	CRDN0068
38	IBC(I)=IT2	CRDN0069
	GO TO 46	CRDN0070
39	IF(IT2-IBC(I)) 41,46,42	CRDN0071
40	IST=IBC(I)	CRDN0072
	NIT=IT2	CRDN0073
	IBC(I)=IT2	CRDN0074

```

41 IF(NIT.NE.IT2) GO TO 40
    ITS=IBC(I+1)
    IBC(I+1)=IST
    IF(ITS.EQ.0) GO TO 46
    IST=ITS
42 CONTINUE
    GO TO 46
43 CONTINUE
    DO 44 I=1,9
    IF(IT2-IBC(I)) 12,45,44
44 CONTINUE
    IBC(10)=0
    GO TO 12
45 DO 451 J=1,9
451 IBC(J)=IBC(J+1)
    GO TO 12
46 DO 47 I=1,3
    J=3*(IT2-1)+I
47 IB(J)=ITEMP(I+3)
    GO TO 12
48 IF(ITEMP)49,49,51
49 DO 50 I=1,10
50 IBC(I)=0
    GO TO 12
51 CONTINUE
    IB(36)=ITEMP(3)
    DO 52 I=1,5
    J=30+I
52 IB(J)=ITEMP(I+3)
    GO TO 12
55 CONTINUE
    IF(ITEMP)56,56,59
56 DO 57 I=1,22
57 IOBR(I)=0
    GO TO 69
59 CONTINUE
    DO 60 I=1,22
60 IOBR(I)=ITEMP(I+1)
69 CONTINUE
    CALL LAYO (OB,NCH,ITEMP )
    GO TO 12
70 CONTINUE
    DO 71 I=1,24
71 IEMP(I)=ITEMP(I)
    CALL LAYO (EMP,NCH,ITEMP )
    GO TO 12
100 RETURN
    END

```

```

CRDN0075
CRDN0076
CRDN0077
CRDN0078
CRDN0079
CRDN0080
CRDN0081
CRDN0082
CRDN0083
CRDN0084
CRDN0085
CRDN0086
CRDN0087
CRDN0088
CRDN0089
CRDN0090
CRDN0091
CRDN0092
CRDN0093
CRDN0094
CRDN0095
CRDN0096
CRDN0097
CRDN0098
CRDN0099
CRDN0100
CRDN0101
CRDN0102
CRDN0103
CRDN0104
CRDN0105
CRDN0106
CRDN0107
CRDN0108
CRDN0109
CRDN0110
CRDN0111
CRDN0112
CRDN0113
CRDN0114
CRDN0115
CRDN0116
CRDN0117
CRDN0118
CRDN0119
CRDN0120
CRDN0121
CRDN0122

```

Subroutine: CBDAT

Purpose: Computes three C-Band measurements (azimuth,
 elevation and range).

Calling Sequence: CALL CBDAT

Common storages used: /DATCOM/

Subroutines required: DMVTRN, DNORM

CBDAT-1

Usage

The equations describing the measurement model are given in Appendix C of Reference 1.

The principal communication between the data subroutines (CBDAT, CBDATP, etc.) with their using programs is through the labelled common DATCOM. The quantities used by each subroutine are described in tabular form in Table 1.

The output array ϕ BS is described in Table 1, below.

COMMON LOCATIONS

COMMON	LOCATION	NAME	DIMENSION	DESCRIPTION
DATCOM	C(139)	STA	d(5)	STA(1-3), receiving station position in B-frame, (km). STA (4-5), refraction constants.
	C(161)	TB2C ϕ	d(18)	TB2C ϕ (1-9), B-frame to C-frame transform at time of measurements. TB2C ϕ (10-18) are not used.
	C(233)	TT2B ϕ	d(9)	Unit North, East, Down vectors at station in B-frame.
	C(269)	XV	d(3)	Spacecraft position in C-frame at time received signal left the spacecraft, (km).
	C(299)	NFRAC		Refraction key: +1, include refraction corrections; +2, omit.
	C(5) C(37) C(69)	ϕ BS		ϕ BS(1), azimuth (rad) ϕ BS(17), elevation (rad) ϕ BS(33), range (km)

CBDAT-2

Table 1

α_1 Azimuth, x-Angle, or Hour Angle		α_2 Elevation, y-Angle, or Declination		ρ Ranging Observable	$\dot{\rho}$ Doppler Observable	COMMENTS
1	17	33	49			
α_1	α_2	ρ	$\dot{\rho}$			Measurements
\uparrow	\uparrow	\uparrow	\uparrow			
$\frac{\partial \alpha_1}{\partial x_v}$	$\frac{\partial \alpha_2}{\partial x_v}$	$\frac{\partial \rho}{\partial x_v}$	$\frac{\partial \dot{\rho}}{\partial x_v}$			Vehicle Position Partials
\downarrow	\downarrow	\downarrow	\downarrow			
\uparrow	\uparrow	\uparrow	\uparrow			Vehicle Velocity Partials
$\frac{\partial \alpha_1}{\partial \dot{x}_v}$	$\frac{\partial \alpha_2}{\partial \dot{x}_v}$	$\frac{\partial \rho}{\partial \dot{x}_v}$	$\frac{\partial \dot{\rho}}{\partial \dot{x}_v}$			
\downarrow	\downarrow	\downarrow	\downarrow			Measurement Bias Partials
$\frac{\partial \alpha_1}{\partial \alpha_1}$	$\frac{\partial \alpha_2}{\partial \alpha_2}$	$\frac{\partial \rho}{\partial \rho}$	$\frac{\partial \dot{\rho}}{\partial \omega_3}$			
$\frac{\partial \alpha_1}{\partial \tau}$	$\frac{\partial \alpha_2}{\partial \tau}$	$\frac{\partial \rho}{\partial \tau}$	$\frac{\partial \dot{\rho}}{\partial \tau}$			Observing Station Clock Bias Partials
$\frac{\partial \alpha_1}{\partial c}$	$\frac{\partial \alpha_2}{\partial c}$	$\frac{\partial \rho}{\partial c}$	$\frac{\partial \dot{\rho}}{\partial \rho}$			Speed of Light Partials
\uparrow	\uparrow	\uparrow	\uparrow			Receiving Station Position Partials
$\frac{\partial \alpha_1}{\partial x_R}$	$\frac{\partial \alpha_2}{\partial x_R}$	$\frac{\partial \rho}{\partial x_R}$	$\frac{\partial \dot{\rho}}{\partial x_R}$			
\downarrow	\downarrow	\downarrow	\downarrow			Transmitting Station Position Partials (When operating in three- way doppler mode)
0	0	0	0			
0	0	48	$\frac{\partial \rho}{\partial x_T}$	64	$\frac{\partial \dot{\rho}}{\partial x_T}$	
16	32	0	\downarrow		\downarrow	

Layout of OBS-Vector in DATCOM. Used by all data and partial subroutines. Only those portions of the OBS-Vector relevant to the data type being computed are loaded by the various data and partial subroutines; (e.g., CBDAT uses only OBS(1), OBS(17), OBS(33)).

```

$IBFTC MC13CE M94,NODD,XR3
CMC13CE C-BAND MEASUREMENTS
SUBROUTINE CBDAT
COMMON /DATCOM/ BIAS(2), OBS(64), FTR, OMEGA
1, SPDLT, STA(10), TAU, TB2CO(18)
2, TB2CT(18), TT2BO(9), TT2BT(9), XV(12)
3, MLT, MODE, MSTA, MTIM
4, NALIGN, NANG, NFRAC
DOUBLE PRECISION RT(3), SX(3), TT2C(9), XS(3)
1, BIAS, CE, OBS, DELTE
2, DNORM, E, F, FTR, OMEGA
3, RAT, RTMG, SE, SPDLT, STA
4, T, TAU, TB2CO, TB2CT, TT2BO
5, TT2BT, XTD, XTE, XTN, XV
C
C DECLARE LIBRARY FUNCTIONS DOUBLE PRECISION TO SATISFY UNIVAC
DOUBLE PRECISION DATAN2,DATAN,DCOS,DSIN,DSQRT
C
EQUIVALENC (SX(1),XTN), (SX(2),XTE), (SX(3),XTD)
C
N,E,D VECTORS IN C FRAME
CALL DMVTRN(TB2CO,TT2BO,TT2C,1,3)
C
STATION VECTOR IN C FRAME
CALL DMVTRN(TB2CO,STA,RT,1,1)
C
RANGE VECTOR IN C FRAME
DO 1 I=1,3
1 XS(I)=XV(I)-RT(I)
C
RANGE VECTOR IN TOPOCENTRIC COORDINATES
CALL DMVTRN(TT2C,XS,SX,2,1)
C
STATION AND RANGE VECTOR MAGNITUDES
RTMG=DNORM(RT)
RAT=DNORM(XS)
C
AZIMUTH, ELEVATION, AND RANGE
ORS(1)=DATAN2(XTE,XTN)
ORS(17)=DATAN(-XTD/DSQRT(XTN*XTN+XTE*XTE))
ORS(33)=RAT
GO TO (2,7),NFRAC
C
REFRACTION CORRECTIONS
C
ELEVATION CORRECTION
2 E=ORS(17)
IF(E.LT..01) GO TO 606
SE=DSIN(E)
CE=DCOS(E)
IF(E.GT..17452393D0) GO TO 5
T=(1.03585796D0-(.01072014D0-(.1279119D-7-.1227363D-7/E)/E)*STA
1(4)*CE/SE
3 F=RTMG/RAT
4 DELTE=T-F*((STA(4)+T*T/.)*CE-T*SE)
GO TO 6
5 DELTE=STA(4)*CE/SF
6 OBS(17)=OBS(17)+DELTE
C
RANGE CORRECTION
606 ORS(33)=ORS(33)+(STA(4)/(STA(5)*DSIN(ORS(17))))*1.D-3
C
AZIMUTH IS NOT CORRECTED FOR REFRACTION
7 CONTINUE
IF(OBS(1).LT.0.D0) ORS(1)=ORS(1)+6.2831853071795864
IF(OBS(1).GT.6.2831853071795864) ORS(1)=ORS(1)-6.2831853071795864
RETURN
END

```

Subroutine: CBDATP

Purpose: Computes three C-Band measurements and their
partial derivatives.

Calling Sequence: CALL CBDATP(R2)

Input and Output

I/φ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
φ	R2	d	S ²	km ²	Slant range squared.

Common storages used: /DATCOM/

Subroutines required: DCR~~OS~~S, DD~~O~~T, DMVTRN, DN~~O~~RM

CBDATP-1

COMMON LOCATIONS

COMMON	LOCATION	NAME	DIMENSION	DESCRIPTION
DATCOM	C(135)	OMEGA	d	Earth rotation rate, (rad/sec).
	C(137)	SPDLT	d	Speed of light, (km/sec).
	C(139)	STA	d(5)	STA(1-3), Receiving station position in B-frame (km). STA(4-5) Refraction constants.
	C(161)	TB2CØ	d(9)	B-frame to C-frame transform at time of measurements.
	C(233)	TT2BØ	d(9)	Unit North, East, Down vectors at station in B-frame.
	C(269)	XV	d(6)	Spacecraft position and velocity in C-frame at time received signal left the spacecraft (km, km/sec).
	C(293)	MLT		Speed of light partial option key*.
	C(295)	MSTA		Station location partial option key*.
	C(296)	MTIM		Station clock partial option key*.
	C(299)	NFRAC		Refraction correction option key*.
				*When option key = 1, option will be included; option will be omitted if key = 2.
	C(5)	ØBS	d(64)	Measurement and partial output vector. See Table 1 in description of CDBAT.

CBDATP-2

```

$IBFTC MC13CD XR3,M94,NODD,LIST
CMC13CD C-BAND MEASUREMENTS AND PARTIAL DERIVATIVES
SUBROUTINE CBDATP(R2)
COMMON /DATCOM/ BIAS(2), OBS(64), FTR, OMEGA
1, SPDLT, STA(10), TAU, TB2CO(18)
2, TB2CT(18), TT2BO(9), TT2RT(9), XV(12)
3, MLT, MODE, MSTA, MTIM
4, NALIGN, NANG, NFRAC
DOUBLE PRECISION DCXS(3), DN(3), DUM(3)
1, EA(3), EN(3), OMG(3)
2, RT(3), SX(3), TT2C(9)
3, XS(6), Z1(3)
DOUBLE PRECISION BIAS, CE, OBS, DDOT
1, DELTE, DEN, DNORM, E, E2
2, E3, F, FTR, OMEGA, RAT
3, RTMG, R2, SE, SPDLT, STA
4, S1, S2, T, TAU, TB2CO
5, TB2CT, TT2BO, TT2RT, XTD, XTE
6, XTN, XV
C
C DECLARE LIBRARY FUNCTIONS DOUBLE PRECISION TO SATISFY UNIVAC
C DOUBLE PRECISION DATAN2,DATAN,DCOS,DSIN,DSQRT
C
EQUIVALENCE (TT2C(1),FN), (TT2C(4),EA), (TT2C(7),DN)
1, (SX(1),XTN), (SX(2),XTE), (SX(3),XTD)
CALL DMVTRN(TB2CO,STA,RT,1,1)
CALL DMVTRN(TB2CO,TT2BO,TT2C,1,3)
DO 1 I=1,3
1 OMG(I)=TB2CO(I+6)*OMEGA
CALL DCROSS(OMG,RT,DUM)
DO 2 I=1,3
XS(I)=XV(I)-RT(I)
2 XS(I+3)=XV(I+3)-DUM(I)
CALL DMVTRN(TT2C,XS,SX,7,1)
RTMG=DNORM(RT)
R2=DDOT(XS,XS)
RAT=DSQRT(R2)
C COMPUTE AZIMUTH, FLEVATION, AND RANGE
ORS(1)=DATAN2(XTE,XTN)
ORS(17)=DATAN(-XTD/DSQRT(XTN*XTN+XTE*XTE))
ORS(33)=RAT
20 E=ORS(17)
IF(E.LT..01) GO TO 56
SE=DSIN(E)
CE=DCOS(F)
IF(E.GE..1745329300) GO TO 5
E2=E*E
E3=E2*E
T=(1.03585796D0-.01072014D0/E+.1279119D-7/E2-.1227363D-7/E3)*STA(4)
1)*CE/SE
3 F=RTMG/RAT
4 DELTE=T-F*((STA(4)+T*T/2.)*CE-T*SE)
GO TO 6
5 DELTE=STA(4)*CE/SE
6 ORS(17)=ORS(17)+DFLTE
C RANGE CORRECTION
56 ORS(33)=ORS(33)+(STA(4)/(STA(5)*DSIN(ORS(17))))*1.D-3
C AZIMUTH IS NOT CORRECTED FOR REFRACTION
60 CONTINUE
IF(ORS(1).LT.0.D0) ORS(1)=ORS(1)+6.2831853071795864
IF(ORS(1).GT.6.2831853071795864) ORS(1)=ORS(1)-6.2831853071795864
C COMPUTE PARTIAL DERIVATIVES
C VEHICLE STATE PARTIALS
CALL DCROSS(DN,XS,DCXS)
S2=DDOT(DCXS,DCXS)
S1=DSQRT(S2)
CALL DCROSS(DCXS,XS,OBS(18))
DO 7 I=1,3
OBS(I+1)=DCXS(I)/S2
ORS(I+17)=OBS(I+17)/(R2*S1)
OBS(I+33)=XS(I)/RAT
ORS(I+4)=0.D0
ORS(I+20)=0.D0
7 ORS(I+36)=0.D0
C MEASUREMENT BIAS PARTIALS

```

	OBS(8)=1.D0	CBDP0075
	OBS(24)=1.D0	CBDP0076
	OBS(40)=1.D0	CBDP0077
	GO TO (8,10),MTIM	CBDP0078
C	OBSERVING STATION TIME BIAS PARTIALS	CBDP0079
8	CALL DCROSS(XS,OMG,Z1)	CBDP0080
	DO 9 I=1,3	CBDP0081
9	Z1(I)=Z1(I)+XS(I+3)	CBDP0082
	OBS(9)=DDOT(OBS(2),Z1)	CBDP0083
	OBS(25)=DDOT(OBS(18),Z1)	CBDP0084
	OBS(41)=DDOT(XS,XS(4))/RAT	CBDP0085
10	GO TO (11,12),MLT	CBDP0086
C	SPEED OF LIGHT PARTIALS	CBDP0087
11	DEN=SPDLT*SPDLT	CBDP0088
	OBS(10)=DDOT(OBS(7),XV(4))*RAT/DEN	CBDP0089
	OBS(26)=DDOT(OBS(18),XV(4))*RAT/DEN	CBDP0090
	OBS(42)=DDOT(XS,XV(4))/DEN-RAT/SPDLT	CBDP0091
12	GO TO (13,16),MSTA	CBDP0092
C	OBSERVING STATION LOCATION ERROR PARTIALS	CBDP0093
13	DO 14 I=2,34,16	CBDP0094
14	CALL DMVTRN(TB2CO,OBS(I),OBS(I+9),2,1)	CBDP0095
	DO 15 I=11,13	CBDP0096
	OBS(I)=-OBS(I)	CBDP0097
	OBS(I+16)=-OBS(I+16)	CBDP0098
15	OBS(I+32)=-OBS(I+32)	CBDP0099
16	RETURN	CBDP0100
	END	

Subroutine: **CBTEST**

Purpose: To read C-Band raw data tapes, decode the data, convert units and, using subroutine P ϕ LYFT, test for outliers and write the edited data in the DCP format on Unit 12.

Calling Sequence: **CALL CBTEST (IERR)**

Input and Output:

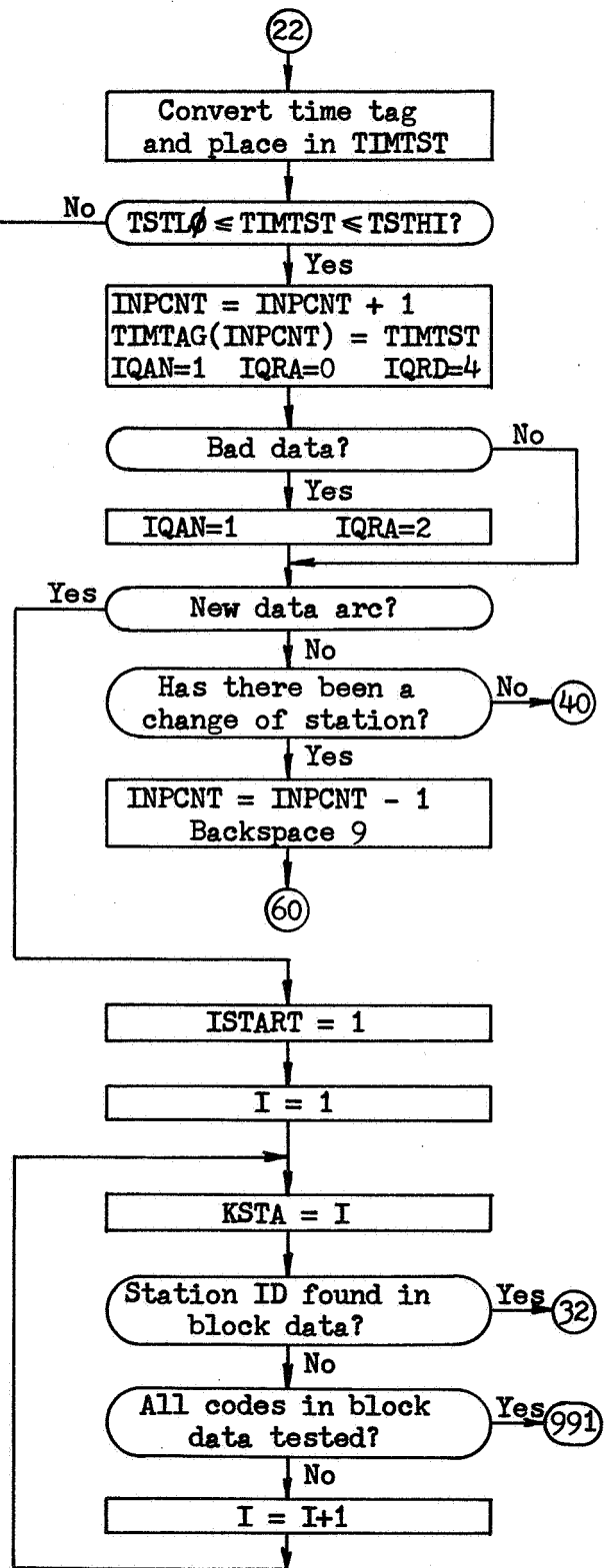
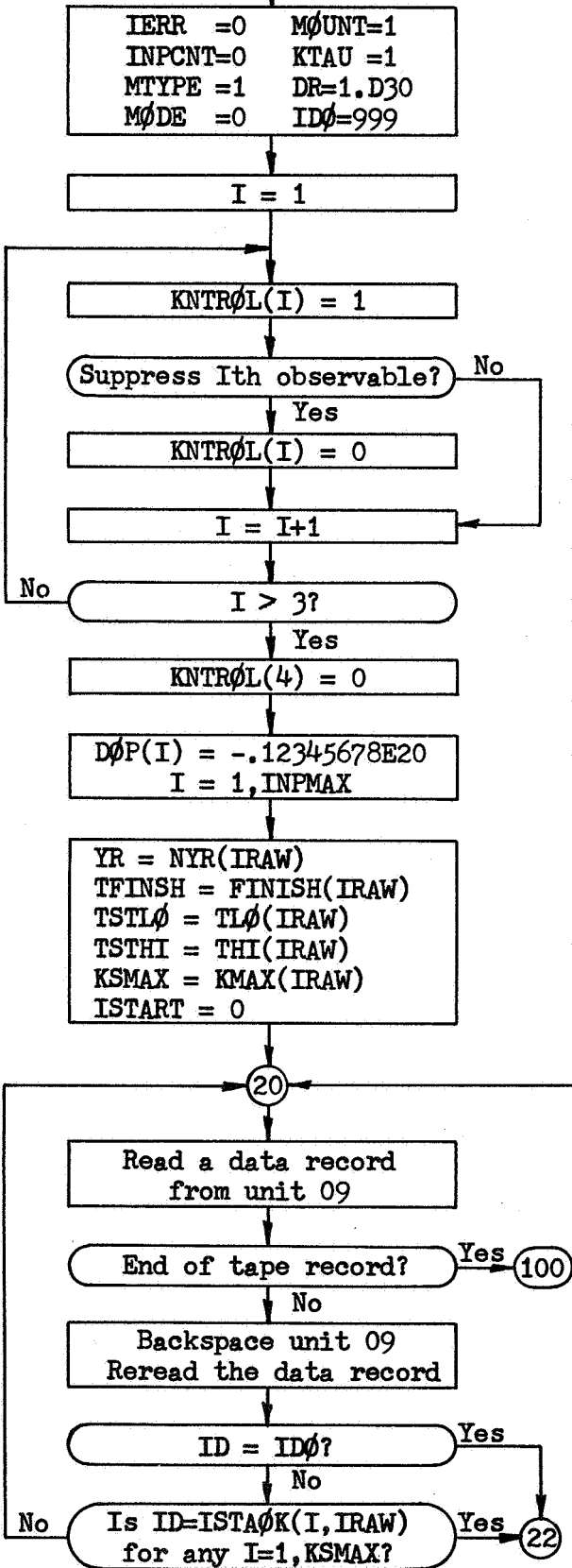
I/ ϕ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
ϕ	IERR				Error flag. Set = 2 by CBTEST if more than 20 stations are accumulated. Set = 0 otherwise.

Common storages used: /TRK ϕ M/, / ϕ UT ϕ M/, /TST ϕ M/, /DAT ϕ M/, /SUM ϕ M/

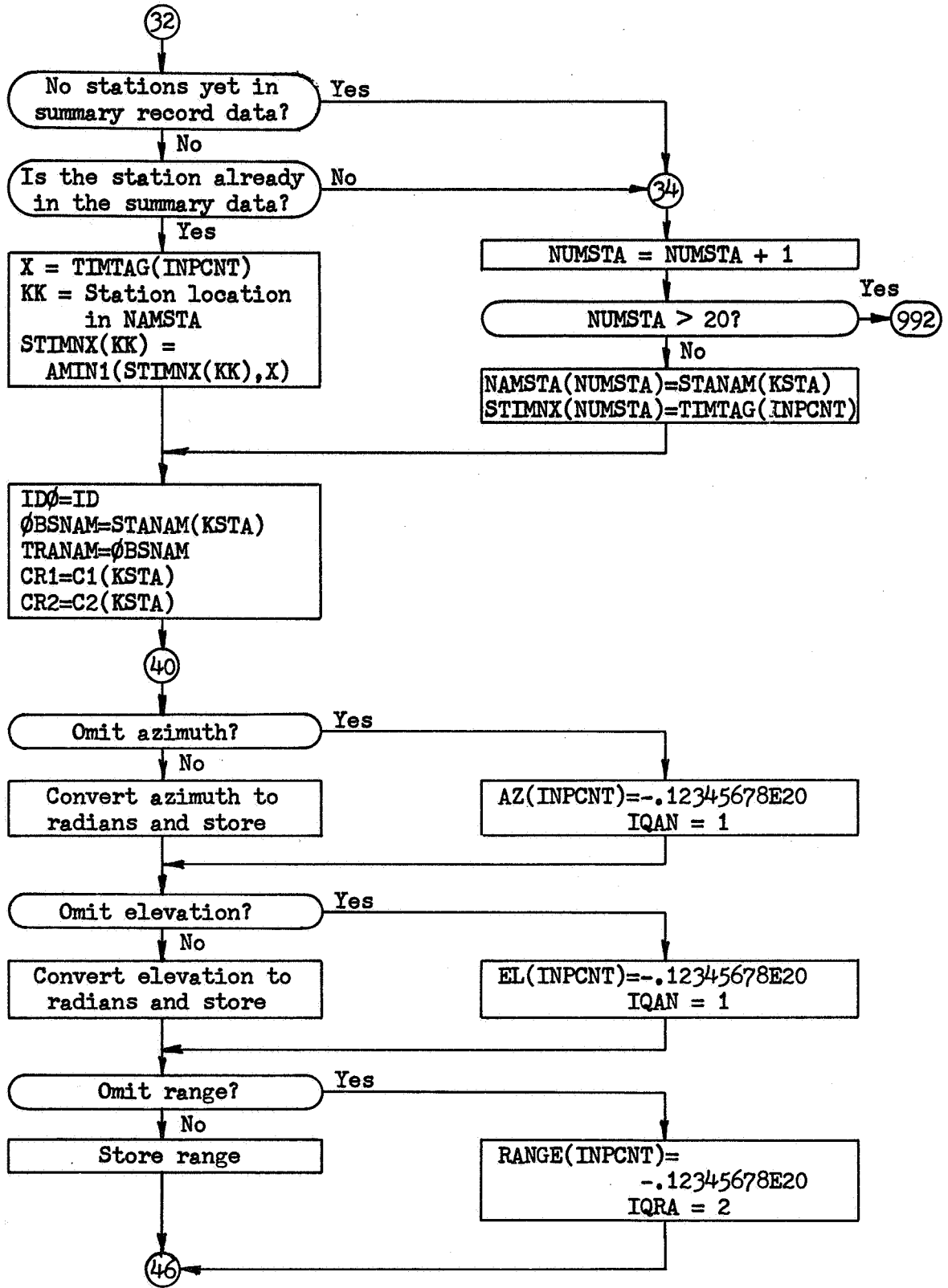
Subroutines required: DATINP, P ϕ LYFT

CBTEST-1

CBTEST



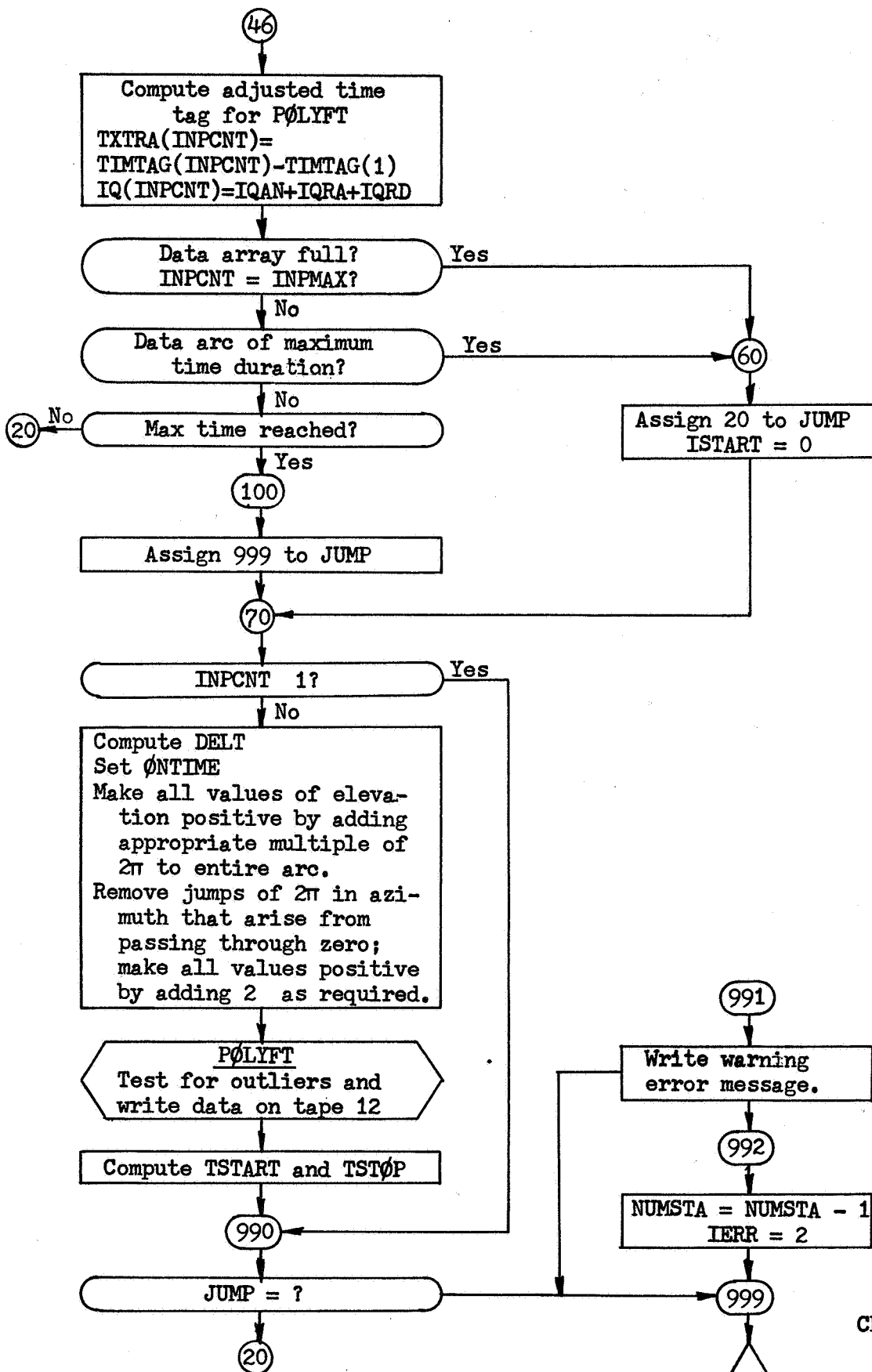
CBTEST-2



CBTEST-3



Space & Re-entry Systems Division



CBTEST-4

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$IBFTC MC134Y M94,NODD,XR3
CMC134Y C-BAND RAW DATA PROCESSOR
SUBROUTINE CBTEST(IERR)
COMMON /TRKCOM/CTRK(700)
DOUBLE PRECISION FTR(50), C1(50), C2(50), BIAS(50), RETR(50)
DIMENSION STANAM(50), KODSTA(50), NALIGN(50), PAIR(50)
EQUIVALENCE (CTRK(1),STANAM), (CTRK(51),KODSTA)
1, (CTRK(101),NALIGN), (CTRK(151),FTR), (CTRK(251),C1)
2, (CTRK(351),C2), (CTRK(451),BIAS), (CTRK(551),RETR)
3, (CTRK(651),PAIR)
C
COMMON /OUTCOM/COU(40)
DOUBLE PRECISION ONTIME, TFIRST, TLAST, TAU, TRF
1, CR1, CR2, DR, BIASF, RATIO
EQUIVALENCE (COU(1),NPAIR), (COU(2),NEOT)
1, (COU(3),OBSNAM), (COU(4),TRANAM), (COU(5),NRCD)
2, (COU(6),NPTBLK), (COU(7),KONT), (COU(8),MTYPE)
3, (COU(9),MOUNT), (COU(10),MODE), (COU(11),DELT)
4, (COU(12),KTAU), (COU(13),ONTIME), (COU(15),TFIRST)
5, (COU(17),TLAST), (COU(19),TAU), (COU(21),TRF)
6, (COU(23),CR1), (COU(25),CR2), (COU(27),DR)
7, (COU(29),BIASF), (COU(31),RATIO), (COU(33),NB1)
8, (COU(34),NB2), (COU(35),NB3), (COU(36),NB4)
C
COMMON /TSTCOM/CTEST(400)
DOUBLE PRECISION FINISH(10), TLO(10), THI(10)
DIMENSION NYR(10), NPTS(10), NSTEP(10), NDEG(10)
1, CSD(4,10), KNTROL(4), IFOMIT(4,10), KMAX(10)
2, ISTAOK(20,10)
EQUIVALENCE (CTEST(1),NYR), (CTEST(11),NPTS)
1, (CTEST(21),NSTEP), (CTEST(31),NDEG), (CTEST(41),CSD)
2, (CTEST(81),IRAW), (CTEST(82),KNTROL), (CTEST(87),FINISH)
3, (CTEST(107),IFOMIT), (CTEST(147),KMAX), (CTEST(157),ISTAOK)
4, (CTEST(357),TLO), (CTEST(377),THI)
C
COMMON /DATCOM/CDAT(2400)
DOUBLE PRECISION TIMTAG(300)
DIMENSION TXTRA(300), IQ(300), AZ(300), EL(300)
1, RANGE(300), DOP(300)
EQUIVALENCE (CDAT(1),TIMTAG), (CDAT(601),TXTRA)
1, (CDAT(901),IQ), (CDAT(1201),AZ), (CDAT(1501),EL)
2, (CDAT(1801),RANGE), (CDAT(2101),DOP)
C
COMMON /SUMCOM/SUMARY(56)
DOUBLE PRECISION TSTART,TSTOP
DIMENSION HEADER(11), STIMNX(20)
REAL NAMSTA(20)
EQUIVALENCE (SUMARY(1),HEADER), (SUMARY(12),NUMSTA)
1, (SUMARY(13),NAMSTA), (SUMARY(33),TSTART), (SUMARY(35),TSTOP)
2, (SUMARY(37),STIMNX)
C
DOUBLE PRECISION TFINSH, TMAX, TIMTST, TSTLO, TSTHI
DIMENSION DI(3)
C
C THESE TWO ARRAYS MUST HAVE THE SAME DIMENSION AS THE ARRAYS IN
C DATCOM. THE DIMENSION IS EQUAL TO THE VARIABLE INPMAX.
C THE VARIABLE INPMAX IS SET BY A DATA STATEMENT
C IN THIS SUBROUTINE.
C TO AVOID TRUNCATING GODDARD SYSTEM DATA FRAMES, INPMAX MUST
C BE A MULTIPLE OF FOUR
C DIMENSION HOLD(300), ITEM(300)
C
C
602 FORMAT(A2)
605 FORMAT(I2,2X,01,3X,F3.0,1X,2F2.0,1X,F6.3,2X,F7.3,2X,F7.3,2X,F12.6)
901 FORMAT(18H0*** STATION CODE,I3,29H DOES NOT EXIST IN BLOCK DATA/
16X,34HPROCESSING OF THIS TAPE ABANDONED./
26X,38HPROGRAM PROCEEDS TO NEXT TAPE, IF ANY.)
902 FORMAT(24H0*** TOO MANY STATIONS./
16X,49HPROGRAM CONTROL PASSES TO TERMINATION OPERATIONS.)
DATA EOTIND/2HED/
DATA INPMAX/300/
DATA TMAX/.432D5/
DATA TEST/-.12345678E20/
IERR=0

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CBTS0001
CBTS0002
CBTS0003
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CBTS0010
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CBTS0055
CBTS0056
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CBTS0058
CBTS0059
CBTS0060
CBTS0061
CBTS0062
CBTS0063
CBTS0064
CBTS0065
CBTS0066
CBTS0067
CBTS0068
CBTS0069
CBTS0070
CBTS0071
CBTS0072
CBTS0073
CBTS0074

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	INPCNT=0	CBTS0075
	MTYPE=1	CBTS0076
	MOUNT=1	CBTS0077
	MODE=0	CBTS0078
	KTAU=1	CBTS0079
	IDO=999	CBTS0080
C	SET RANGE AMBIGUITY LARGE	CBTS0081
C	IF RANGE REALLY DOES RECYCLE, YOU WILL CHANGE THIS	CBTS0082
	DR=1.D30	CBTS0083
	DO 1 I=1,3	CBTS0084
	KNTROL(I)=1	CBTS0085
	IF(IFOMIT(I,IRAW).NE.0) KNTROL(I)=0	CBTS0086
1	CONTINUE	CBTS0087
	KNTROL(4)=0	CBTS0088
	DO 2 I=1,INPMAX	CBTS0089
2	DOP(I)=TEST	CBTS0090
	YR=NYP(IRAW)	CBTS0091
	TFINSH=FINISH(IRAW)	CBTS0092
	TSTLO=TLO(IRAW)	CBTS0093
	TSTHI=THI(IRAW)	CBTS0094
	KSMAX=KMAX(IRAW)	CBTS0095
	ISTART=0	CBTS0096
C	TEST FOR END OF TAPE INDICATOR	CBTS0097
20	READ(9,602) ARCIND	CBTS0098
	IF(ARCIND.EQ.EOTIND) GO TO 100	CBTS0099
	BACKSPACE 9	CBTS0100
C	READ A LINE OF DATA	CBTS0101
	READ(9,605) ID,IDC,DAY,HR,XMIN,SEC,AZX,ELX,RAX	CBTS0102
C	FIND IF STATION NUMBER IS UNCHANGED	CBTS0103
	IF(ID.EQ.IDO) GO TO 22	CBTS0124
C	TEST IF STATION ID VALID	CBTS0125
	DO 21 I=1,KSMAX	CBTS0126
	IF(ID.EQ.ISTAOK(I,IRAW)) GO TO 22	CBTS0127
21	CONTINUE	CBTS0128
	GO TO 20	CBTS0129
22	CONTINUE	CBTS0130
C	DECODE DATA KEYS AND CONVERT TIME TAG	CBTS0131
	DI(1)=YR*100.+1.	CBTS0132
	DI(2)=(DAY*100.+HR)*100.+XMIN	CBTS0133
	DI(3)=SEC	CBTS0134
	CALL DATINP(DI,TIMTST)	CBTS0135
C	TEST TIME TAG FOR VALID RANGE OF TIMES	CBTS0136
	IF(TIMTST.GE.TSTLO.AND.TIMTST.LE.TSTHI) GO TO 220	CBTS0137
	GO TO 20	CBTS0138
220	INPCNT=INPCNT+1	CBTS0139
	IQ(INPCNT)=0	CBTS0140
	TIMTAG(INPCNT)=TIMTST	CBTS0141
	IQAN=0	CBTS0142
	IQRA=0	CBTS0143
	IQRD=4	CBTS0144
	IF(IDC.NE.0) GO TO 23	CBTS0145
	IQAN=1	CBTS0146
	IQRA=2	CBTS0147
23	IF(ISTART.EQ.0) GO TO 30	CBTS0148
C	IF STATION CHANGE, A NEW DATA ARC IS FORCED	CBTS0149
	IF(ID.EQ.IDO) GO TO 40	CBTS0150
	INPCNT=INPCNT-1	CBTS0151
	BACKSPACE 9	CBTS0152
	GO TO 60	CBTS0153
C	NEW DATA ARC	CBTS0154
30	ISTART=1	CBTS0155
C	FIND STATION ID IN BLOCK DATA	CBTS0156
	DO 31 I=1,50	CBTS0157
	KSTA=I	CBTS0158
	IF(ID.EQ.KODSTA(I)) GO TO 32	CBTS0159
31	CONTINUE	CBTS0160
	GO TO 991	CBTS0161
32	IF(NUMSTA.EQ.0) GO TO 34	CBTS0162
C	IS STATION NAME ALREADY IN SUMMARY DATA	CBTS0163
	DO 33 I=1,NUMSTA	CBTS0164
	KK=I	CBTS0165
	IF(NAMSTA(I).EQ.STANAM(KSTA)) GO TO 35	CBTS0166
33	CONTINUE	CBTS0167
C	NEW STATION NAME	CBTS0168
34	NUMSTA=NUMSTA+1	CBTS0169

	IF(NUMSTA.GT.20) GO TO 992	CBTS0170
	NAMSTA(NUMSTA)=STANAM(KSTA)	CBTS0171
	STIMNX(NUMSTA)=TIMTAG(INPCNT)	CBTS0172
	GO TO 36	CBTS0173
C	OLD STATION NAME	CBTS0174
C	UPDATE STATION FIRST ON TIME	CBTS0175
	35 X=TIMTAG(INPCNT)	CBTS0176
	STIMNX(KK)=AMIN1(STIMNX(KK),X)	CBTS0177
	36 IDO=ID	CBTS0178
	OBSNAM=STANAM(KSTA)	CBTS0179
	TRANAM=OBSNAM	CBTS0180
	CR1=C1(KSTA)	CBTS0181
	CR2=C2(KSTA)	CBTS0182
	40 CONTINUE	CBTS0183
C	LOAD ROW OF DATA ARRAY	CBTS0184
	IF(KNTR0L(1).NE.0) GO TO 41	CBTS0185
	AZ(INPCNT)=TEST	CBTS0186
	IQAN=1	CBTS0187
	GO TO 42	CBTS0188
	41 AZ(INPCNT)=AZX*.0174532926	CBTS0189
	42 IF(KNTR0L(2).NE.0) GO TO 43	CBTS0190
	EL(INPCNT)=TEST	CBTS0191
	IQAN=1	CBTS0192
	GO TO 44	CBTS0193
	43 EL(INPCNT)=ELX*.0174532926	CBTS0194
	44 IF(KNTR0L(3).NE.0.AND.RAX.NE.0.) GO TO 45	CBTS0195
	RANGE(INPCNT)=TEST	CBTS0196
	IQRA=2	CBTS0197
	GO TO 46	CBTS0198
	45 RANGE(INPCNT)=RAX	CBTS0199
	46 TXTRA(INPCNT)=TIMTAG(INPCNT)-TIMTAG(1)	CBTS0200
	IQ(INPCNT)=IQAN+IQRA+IQRD	CBTS0201
	IF(INPCNT.EQ.INPMAX) GO TO 60	CBTS0202
	IF(TXTRA(INPCNT).GE.TMAX) GO TO 60	CBTS0203
	IF(TIMTAG(INPCNT).GE.TFINSH) GO TO 100	CBTS0204
	GO TO 20	CBTS0205
	60 ASSIGN 20 TO JUMP	CBTS0206
	ISTART=0	CBTS0207
	GO TO 70	CBTS0208
	100 ASSIGN 999 TO JUMP	CBTS0209
	70 IF(INPCNT.LE.1) GO TO 990	CBTS0210
C	FROM HERE TO EFN 705 IS ALL TO FIND THE MOST COMMON TIME INTERVAL	CBTS0211
C	BETWEEN MEASUREMENTS. THIS WILL BE PUT INTO DELT	CBTS0212
	MANY=1	CBTS0213
	ITEM(1)=1	CBTS0214
	HOLD(1)=TIMTAG(2)-TIMTAG(1)	CBTS0215
	IF(INPCNT.LE.2) GO TO 704	CBTS0216
	DO 703 I=3,INPCNT	CBTS0217
	HERE=TIMTAG(I)-TIMTAG(I-1)	CBTS0218
	DO 701 J=1,MANY	CBTS0219
	IF(HERE.EQ.HOLD(J)) GO TO 702	CBTS0220
	701 CONTINUE	CBTS0221
	MANY=MANY+1	CBTS0222
	HOLD(MANY)=HERE	CBTS0223
	ITEM(MANY)=1	CBTS0224
	GO TO 703	CBTS0225
	702 ITEM(J)=ITEM(J)+1	CBTS0226
	703 CONTINUE	CBTS0227
	704 MOST=1	CBTS0228
	ISRCH=1	CBTS0229
	DO 705 I=1,MANY	CBTS0230
	IF(MOST.GE.ITEM(I)) GO TO 705	CBTS0231
	MOST=ITEM(I)	CBTS0232
	ISRCH=I	CBTS0233
	705 CONTINUE	CBTS0234
	DELT=HOLD(ISRCH)	CBTS0235
	ONTIME=TIMTAG(1)	CBTS0236
C	MAKE ALL ELEVATIONS POSITIVE IF NECESSARY	CBTS0237
C	BUT RETAIN CONTINUITY	CBTS0238
	DO 710 I=1,INPCNT	CBTS0239
	IF(EL(I).EQ.TEST) GO TO 710	CBTS0240
	IF(EL(I).GE.0.) GO TO 710	CBTS0241
	DO 71 L=1,INPCNT	CBTS0242
	IF(EL(L).EQ.TEST) GO TO 71	CBTS0243
	EL(L)=EL(L)+6.28318531	CBTS0244

71	CONTINUE	CBTS0245
710	CONTINUE	CBTS0246
C	SMOOTH OUT THE AZIMUTH DISCONTINUITIES AT ZERO=TWO PI	CBTS0247
	DO 72 I=1,INPCNT	CBTS0248
	KK=I+1	CBTS0249
	IF(AZ(I).NE.TEST) GO TO 73	CBTS0250
72	CONTINUE	CBTS0251
	GO TO 79	CBTS0252
73	IF(KK.GT.INPCNT) GO TO 79	CBTS0253
	KHOLD=KK-1.	CBTS0254
	DO 78 I=KK,INPCNT	CBTS0255
	IF(AZ(I).EQ.TEST) GO TO 78	CBTS0256
	DIFF=AZ(I)-AZ(KHOLD)	CBTS0257
	IF(ABS(DIFF).LT.3.14159266) GO TO 77	CBTS0258
	IF(DIFF.LT.0.) GO TO 74	CBTS0259
	II=KK	CBTS0260
	LL=KHOLD	CBTS0261
	GO TO 75	CBTS0262
74	II=I	CBTS0263
	LL=INPCNT	CBTS0264
75	DO 76 M=II,LL	CBTS0265
	IF(AZ(M).EQ.TEST) GO TO 76	CBTS0266
	AZ(M)=AZ(M)+6.28318531	CBTS0267
76	CONTINUE	CBTS0268
77	KHOLD=I	CBTS0269
78	CONTINUE	CBTS0270
79	CONTINUE	CBTS0271
C	DO POLYNOMIAL FITS, TEST FOR OUTLIERS	CBTS0272
C	PUT DATA ON TAPE 12	CBTS0273
	CALL POLYFT(INPCNT)	CBTS0274
	TSTART=DMIN1(TSTART,TIMTAG(1))	CBTS0275
	TSTOP=DMAX1(TSTOP,TIMTAG(INPCNT))	CBTS0276
990	INPCNT=0	CBTS0277
	GO TO JUMP,(20,999)	CBTS0278
991	WRITE(6,901)ID	CBTS0279
	GO TO 999	CBTS0280
992	WRITE(6,902)	CBTS0281
	NUMSTA=NUMSTA-1	CBTS0282
	IERR=2	CBTS0283
999	RETURN	CBTS0284
	END	CBTS0285

Subroutine: CØVØUT

Purpose: To output the normalized covariance matrix in the base coordinate system and, optionally, either Darboux or local tangent plane coordinates.

Calling Sequence: CALL CØVØUT (NNP,X,LD)

Input and Output

I/ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	NNP				Dimension of upper-left portion of P to be output.
I	X	d(6)		km km/sec	Vehicle position and velocity in base coordinate system (body centered).
I	LD				LD=1, P will (also) be output in Darboux, <u>or</u> ; LD=2, P will be output in local tangent plane.

Common Storages used: //248 cells,/ES1CØM/

Subroutines Required: DTRANP, DTRDB, DMVTRN

CØVØUT-1

Organization of ILØC and KLØC

These index arrays are used throughout the Differential Correction Program to identify the elements of the state vector. The allowed state variables are divided logically into type groups; each group has its identifying code number as follows:

<u>Code</u>	<u>Description</u>	<u>Maximum Number of Elements</u>
1	Spacecraft position and velocity	6
2	Planetary mu's and ephemeris parameters	14
3	Solar pressure, drag, and venting	4
4	Ephemeris time to universal time conversion	2
5	Earth gravitational harmonics	24
6	Moon gravitational harmonics	24
7	Extra body gravitational harmonics	24
100+k	Tracker location and measurement bias for station k. (k is an internal number assigned by the D.C.P.)	10

The values of KLØC are upper loaded and are considered in pairs. The first value of a pair is one of the type group codes described above; the second value of a pair is the number of variables from that type group that are included in the state. The specific variables are identified by values in the ILØC array. The following example shows how the ILØC and KLØC arrays are built up, and should make their use clear:

Suppose the state is to consist, first of all, of all six spacecraft position and velocity elements. Then $KLØC(1) = 1$, identifying the spacecraft group; $KLØC(2) = 6$, indicating a total of six spacecraft elements. Then the first 6 elements of ILØC are the identifying numbers of the elements chosen from the spacecraft group, in this case, the numbers 1-6. Thus, having inserted zeros in the unused locations:

CØVØUT-2

	KLØC		ILØC	
1	$\begin{pmatrix} 1 \\ 6 \\ 0 \\ 0 \\ \cdot \\ \cdot \\ \cdot \\ 0 \end{pmatrix}$	1	$\begin{pmatrix} 1 \\ 2 \\ 3 \\ 4 \\ 5 \\ 6 \\ 0 \\ 0 \\ \cdot \\ \cdot \\ \cdot \\ 0 \end{pmatrix}$	
54				
				30

Suppose we wish also the mu's of the earth and moon to be included in the state. Then KLØC(3) = 2, identifying the planetary mu group; KLØC(4) = 2, indicating two elements from that group; ILØC(7) = 3, indicating earth mu; and ILØC(8) = 11, indicating moon mu. Thus, we have:

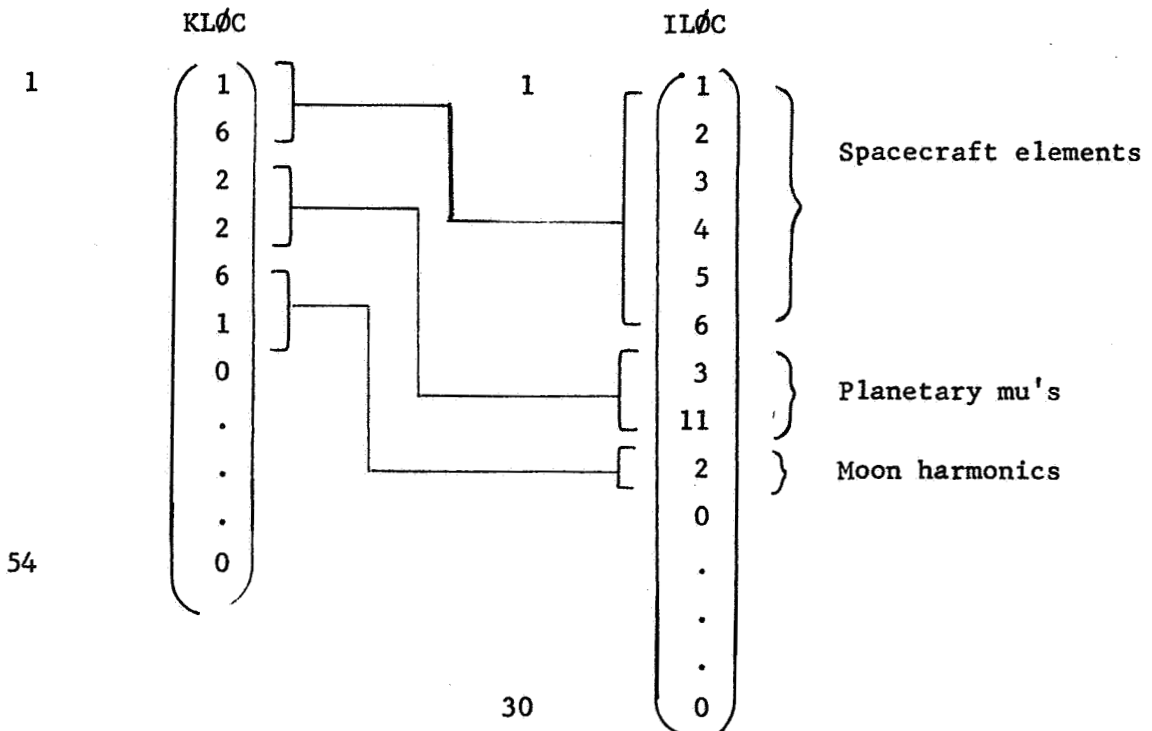
	KLØC		ILØC					
1	$\begin{pmatrix} 1 \\ 6 \\ 2 \\ 2 \\ 0 \\ \cdot \\ \cdot \\ \cdot \\ 0 \end{pmatrix}$		$\begin{pmatrix} 1 \\ 2 \\ 3 \\ 4 \\ 5 \\ 6 \end{pmatrix}$	} Spacecraft elements				
					$\begin{pmatrix} 3 \\ 11 \\ 0 \\ 0 \\ \cdot \\ \cdot \\ \cdot \\ 0 \end{pmatrix}$	} Planetary mu elements		
54								

CØVØUT-3

Suppose we wish also to include one gravitational harmonic for the moon, namely, $J_{3,0}$.

Then, $KL\emptyset C(5) = 6$, indicating the moon harmonic group; $KL\emptyset C(6) = 1$, indicating only one element from that group, and $IL\emptyset C(9) = 2$, identifying the second element from that group, viz., $J_{3,0}$.

Then, we have, showing the correspondence,



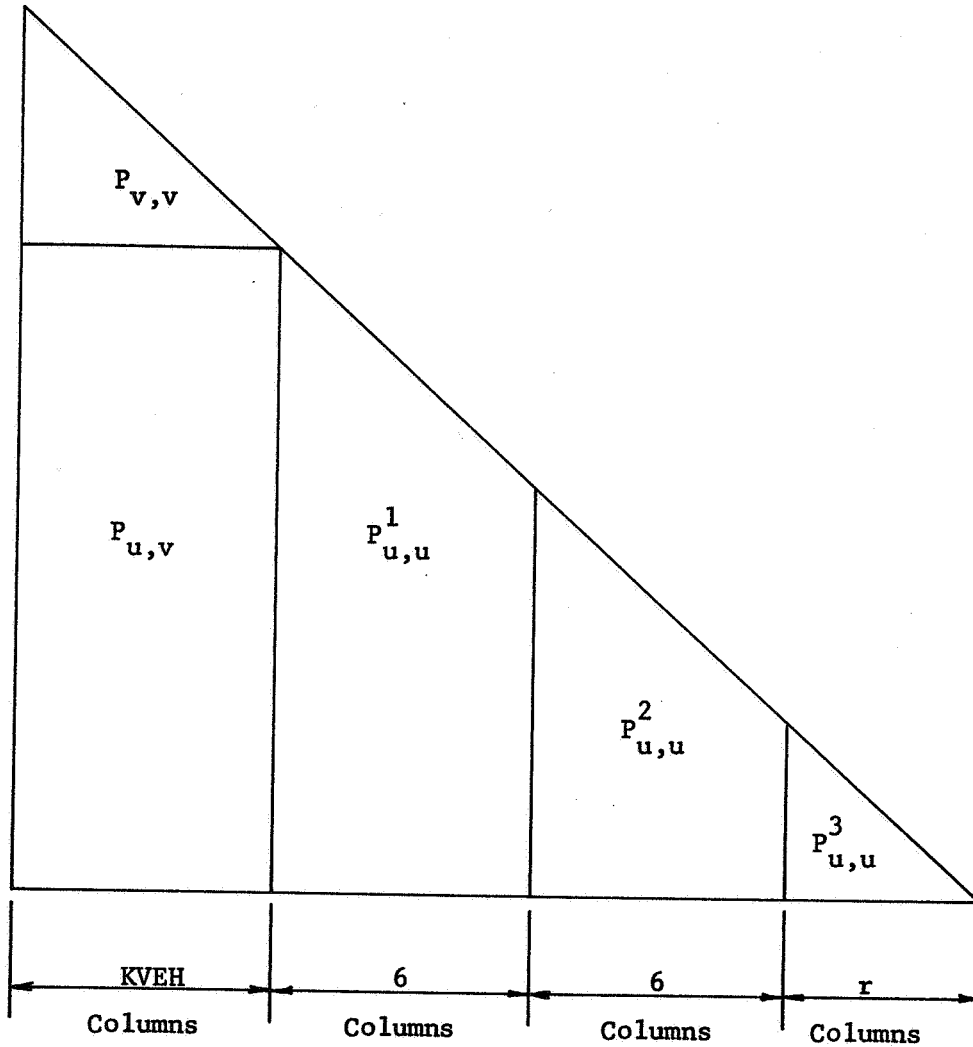
CØVØUT-4

Output Format for Normalized Covariance Matrix

Since the covariance matrix is symmetric, it is unnecessary to display the entire matrix; CØVØUT prints out the lower-left triangular part of the covariance matrix. Let the matrix be considered as partitioned as shown below:

$P_{v,v}$	$P_{v,u}$
$P_{u,v}$	$P_{u,u}$

The matrix $P_{v,v}$ is concerned exclusively with spacecraft position and velocity, $P_{u,u}$ exclusively with non-spacecraft elements of the state vector, and $P_{u,v}$ and $P_{v,u}$ show correlations between these two groups. Further divide the lower-left triangular portion of P as follows:

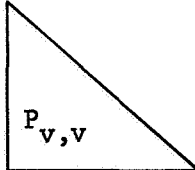


If the matrix is $n \times n$ in size, and $KVEH$ is the number of spacecraft position and velocity elements in the state, then $r = (n - KVEH) \text{ modulo } 6$. Then the matrix is printed out in the following order:

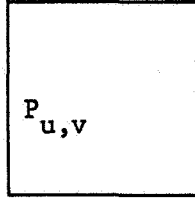
CØVØUT-6



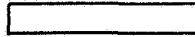
Standard deviations from $P_{v,v}$ coordinates of estimate



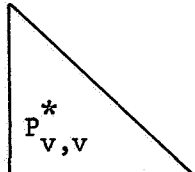
Normalized $P_{v,v}$ in coordinates of estimate



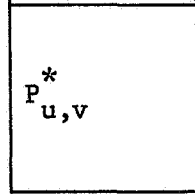
Normalized $P_{u,v}$ in coordinates of estimate



Standard deviations from $P_{v,v}$ after transformation into alternate coordinates (Darboux or tangent plane)



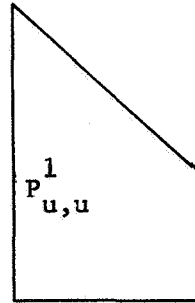
Normalized $P_{v,v}$ transformed into alternate coordinates



Normalized $P_{u,v}$ in alternate coordinates



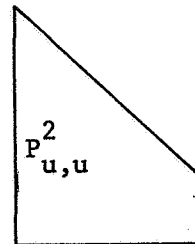
Standard deviations from $P_{u,u}^1$



Normalized $P_{u,u}^1$



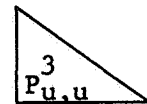
Standard deviations from $P_{u,u}^2$



Normalized $P_{u,u}^2$



Standard deviations from $P_{u,u}^3$



Normalized $P_{u,u}^3$

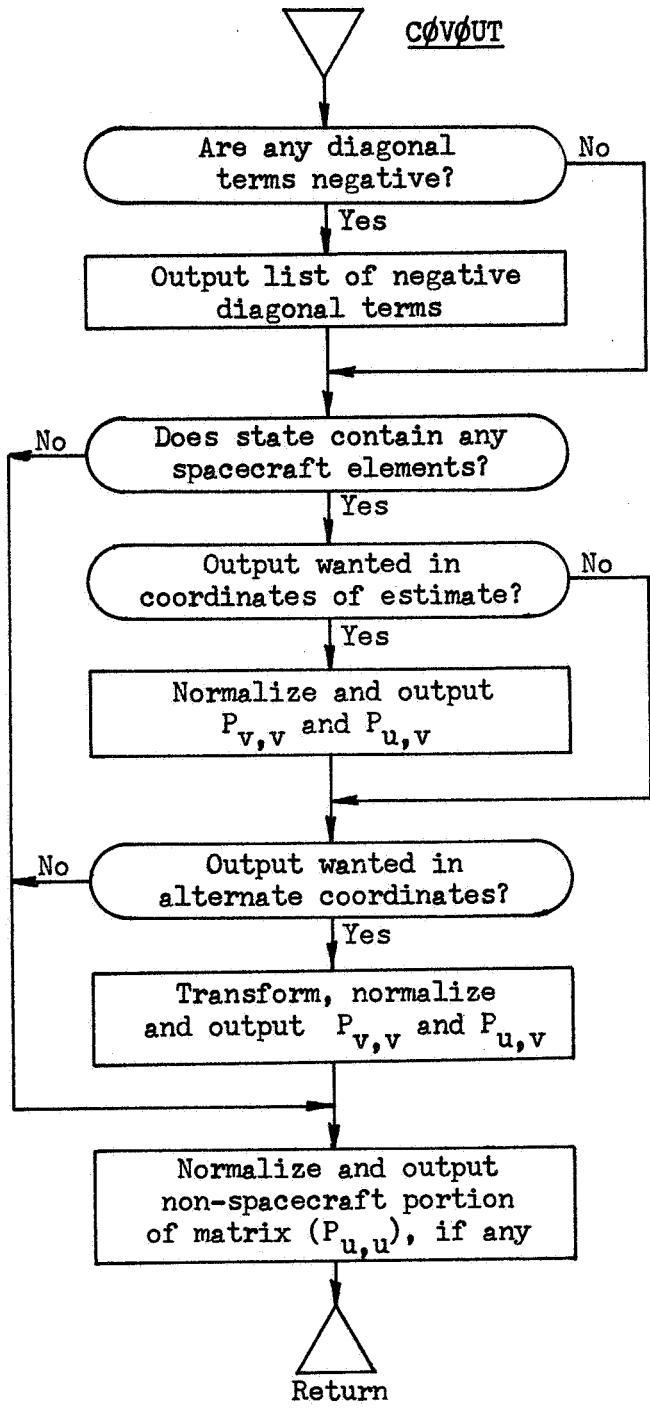
The first block of output will be KVEH Columns wide, one Column for each spacecraft position or velocity element included in the state. The second block of output, representing the first block expressed in alternate coordinates, will be either three or six Columns wide. If the spacecraft portion of the state consists of only position elements or only velocity elements, the transformation will cause the output to occupy three Columns. If the spacecraft portion of the state consists of at least one position element and at least one velocity element, the transformation will cause the output to occupy six Columns.

The remainder of the output is printed six Columns at a time. (Obviously the last block may occupy fewer than six Columns.)

Simplified CØVØUT

The DCP is restricted in the selection of state elements in that either all position and velocity components must be included or all must be omitted. A simplified CØVØUT (deck MC13CV) which uses this restriction to simplify and contract the subroutine is provided. It is completely interchangeable with the complete version (MC13CØ).

CØVØUT-8



CØVØUT-9

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$IBFTC MC13CO XR3,M94,NODD,LIST
SUBROUTINE COVOUT (NNP,X,LD)
C   OUTPUTS NORMALIZED COVARIANCE MATRIX
DOUBLE PRECISION X(6) ,DSQRT
COMMON /ESICOM/PNEW(30,30),STNAMN(30),TRAKER(30)
1      ,ITRETN (30),ILOCN (30),KLOCN (54)
2      ,NSN (12,20)
C   DOUBLE PRECISION PNEW
COMMON D,A,PD,PP,T
DOUBLE PRECISION D(34) ,A(3,3) ,PD(6,6) ,PP(6,6) ,T(3,3)
C   DIMENSION BN(9)
DATA BN/2H N,2H V,2H W,2HND,2HVD,2HWD
1      ,6H ,6HODARBO,6HOTANGT/
C   600 FORMAT(1H0,51X,17HCOVARIANCE MATRIX)
C   601 FORMAT(1H0,37X,44HTHE FOLLOWING STATES HAVE NEGATIVE DIAGONALS//47
1X,5HSTATE,10X,8HVARIANCE)
C   602 FORMAT(41X,A6,1X,A3,D22.8)
C   603 FORMAT(1H0,7X,6(12X,A6))
C   604 FORMAT(A6,6(15X,A3))
C   605 FORMAT(8X,3HSTD,1X,6D18.8)
C   606 FORMAT(1X,A6,1X,A3,4X,F12.9,5(6X,F12.9))
C   607 FORMAT(1H )
C   OUTPUT WILL APPEAR IN COORDINATES OF ESTIMATE
C   IF LD.NE.0 OUTPUT WILL (ALSO) APPEAR IN ALTERNATE COORDINATES
C   LD=1, OUTPUT IN DARBOUX
C   LD=2, OUTPUT IN LOCAL TANGENT PLANE
C
WRITE(6,600)
NP=NNP
KVEH=0
KDEL=0
KP=1
DO 2 I=1,NP
IF(PNEW(I,I).GE.0.) GO TO 2
IF(KP.EQ.0) GO TO 1
WRITE(6,601)
KP=0
1 WRITE(6,602) TRAKER(I),STNAMN(I),PNEW(I,I)
2 D(I)=DSQRT(DABS(PNEW(I,I)))
IF(KLOCN(1).NE.1) GO TO 20
C   OUTPUT PORTION OF P MATRIX ASSOCIATED WITH SPACECRAFT
C   OUTPUT IN COORDINATES OF ESTIMATE
KVEH=KLOCN(2)
DO 3 I=1,KVEH
3 PD(I,I)=1.
DO 4 J=2,KVEH
DO 4 I=J,KVEH
4 PD(I,J-1)=PNEW(I,J-1)/D(I)/D(J-1)
WRITE(6,607)
WRITE(6,604) BN(7),(STNAMN(I),I=1,KVEH)
WRITE(6,605) (D(I),I=1,KVEH)
WRITE(6,607)
DO 5 I=1,KVEH
5 WRITE(6,606) TRAKER(I),STNAMN(I),(PD(I,J),J=1,I)
IF(KVEH.EQ.NP) GO TO 8
KP=KVEH+1
DO 7 I=KP,NP
DO 6 J=1,KVEH
6 PP(J,1)=PNEW(I,J)/D(I)/D(J)
7 WRITE(6,606) TRAKER(I),STNAMN(I),(PP(J,1),J=1,KVEH)
8 IF(LD.EQ.0) GO TO 20
C   OUTPUT IN DARBOUX OR LOCAL TANGENT PLANE
KKJ=0
KP=0
KV=0
KVEH=3
K=KLOCN(2)
DO 9 I=1,K
IF(ILOCN(I).LE.3) KP=1
9 IF(ILOCN(I).GE.4) KV=1
IF(KP+KV.EQ.2) KVEH=6

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COVT0001
COVT0002
COVT0003
COVT0004
COVT0005
COVT0006
COVT0007
COVT0008
COVT0009
COVT0010
COVT0011
COVT0012
COVT0013
COVT0014
COVT0015
COVT0016
COVT0017
COVT0018
COVT0019
COVT0020
COVT0021
COVT0022
COVT0023
COVT0024
COVT0025
COVT0026
COVT0027
COVT0028
COVT0029
COVT0030
COVT0031
COVT0032
COVT0033
COVT0034
COVT0035
COVT0036
COVT0037
COVT0038
COVT0039
COVT0040
COVT0041
COVT0042
COVT0043
COVT0044
COVT0045
COVT0046
COVT0047
COVT0048
COVT0049
COVT0050
COVT0051
COVT0052
COVT0053
COVT0054
COVT0055
COVT0056
COVT0057
COVT0058
COVT0059
COVT0060
COVT0061
COVT0062
COVT0063
COVT0064
COVT0065
COVT0066
COVT0067
COVT0068
COVT0069
COVT0070
COVT0071
COVT0072
COVT0073
COVT0074

```

IF(KP.EQ.0.AND.KV.EQ.1) KKJ=3	COVT0075
KDEL=KVEH-K	COVT0076
NP=NP+KDEL	COVT0077
DO 10 I=1,6	COVT0078
DO 10 J=1,6	COVT0079
10 PP(I,J)=0.	COVT0080
DO 11 I=1,K	COVT0081
KROW=ILOCN(I)	COVT0082
DO 11 J=1,K	COVT0083
KCOL=ILOCN(J)	COVT0084
11 PP(KROW,KCOL)=PNEW(I,J)	COVT0085
CALL DTRDB(X,X(4),T,LD)	COVT0086
DO 12 I=1,3	COVT0087
DO 12 J=1,3	COVT0088
12 A(I,J)=T(J,I)	COVT0089
CALL DTRANP(PP,6,A,A,PD,0)	COVT0090
DO 13 I=1,KVEH	COVT0091
D(I)=DSQRT(DABS(PD(I,I)))	COVT0092
13 PD(I,I)=1.	COVT0093
DO 14 I=2,KVEH	COVT0094
DO 14 J=I,KVEH	COVT0095
14 PD(J,I-1)=PD(J,I-1)/D(J)/D(I-1)	COVT0096
M=KKJ+1	COVT0097
M1=KKJ+KVEH	COVT0098
WRITE(6,604) BN(LD+7),(BN(I),I=M,M1)	COVT0099
WRITE(6,605) (D(I),I=1,KVEH)	COVT0100
WRITE(6,607)	COVT0101
DO 15 I=1,KVEH	COVT0102
M=KKJ+I	COVT0103
15 WRITE(6,606) TRAKER(1),BN(M),(PD(I,J),J=1,I)	COVT0104
IF(KVEH.EQ.NP) GO TO 99	COVT0105
KP=KVEH+1	COVT0106
DO 19 I=KP,NP	COVT0107
KROW=I-KDEL	COVT0108
D(I)=DSQRT(DABS(PNEW(KROW,KROW)))	COVT0109
DO 16 J=1,KVEH	COVT0110
16 PP(J,2)=0.	COVT0111
DO 17 J=1,K	COVT0112
KCOL=ILOCN(J)	COVT0113
17 PP(KCOL,2)=PNEW(KROW,J)	COVT0114
CALL DMVTRN(A,PP(1,2),PP,1,1)	COVT0115
IF(KVEH.EQ.6) CALL DMVTRN(A,PP(4,2),PP(4,1),1,1)	COVT0116
DO 18 J=1,KVEH	COVT0117
18 PP(J,1)=PP(J,1)/D(I)/D(J)	COVT0118
19 WRITE(6,606) TRAKER(KROW),STNAMN(KROW),(PP(J,1),J=1,KVEH)	COVT0119
20 CONTINUE	COVT0120
C OUTPUT PORTION NOT ASSOCIATED WITH SPACECRAFT	COVT0121
K=KVEH	COVT0122
21 IF(NP.LE.K) GO TO 99	COVT0123
KV=NP-K	COVT0124
JJ=6	COVT0125
IF(KV.LT.6) JJ=KV	COVT0126
DO 23 I=1,JJ	COVT0127
L=K+I	COVT0128
L1=L-KDEL	COVT0129
DO 22 J=1,I	COVT0130
M=K+J	COVT0131
M1=M-KDEL	COVT0132
22 PD(I,J)=PNEW(L1,M1)/D(L)/D(M)	COVT0133
23 PD(I,I)=1.	COVT0134
KV=L-JJ+1	COVT0135
KKJ=KV-KDEL	COVT0136
LLJ=L-KDEL	COVT0137
WRITE(6,603) (TRAKER(I),I=KKJ,LLJ)	COVT0138
WRITE(6,604) BN(7),(STNAMN(I),I=KKJ,LLJ)	COVT0139
WRITE(6,605) (D(I),I=KV,L)	COVT0140
WRITE(6,607)	COVT0141
KV=KV-1	COVT0142
DO 24 I=1,JJ	COVT0143
KP=KV+I	COVT0144
KKJ=KP-KDEL	COVT0145
24 WRITE(6,606) TRAKER(KKJ),STNAMN(KKJ),(PD(I,J),J=1,I)	COVT0146
IF(NP.EQ.KP) GO TO 99	COVT0147
KP=KP+1	COVT0148
DO 26 I=KP,NP	COVT0149

```
I1=I-KDEL
DO 25 J=1, JJ
L=KV+J
L1=L-KDEL
25 PP(J,1)=PNEW(L1,I1)/D(L)/D(I)
26 WRITE(6,606) TRAKER(I1),STNAMN(I1),(PP(J,1),J=1, JJ)
K=K+JJ
GO TO 21
99 RETURN
END
```

```
COVT0150
COVT0151
COVT0152
COVT0153
COVT0154
COVT0155
COVT0156
COVT0157
COVT0158
```

```

$IBFTC MC13CV XR3,M94,NODD,LIST
SUBROUTINE COVOUT (NNP,X,LD)
C   OUTPUTS NORMALIZED COVARIANCE MATRIX -- SIMPLIFIED VERSION
C   ASSUMES EITHER A SIX-ELEMENT OR ZERO-ELEMENT VEHICLE STATE
DOUBLE PRECISION X(6),DSQRT
COMMON /ESICOM/PNEW(30,30),STNAMN(30),TRAKER(30)
1   ,ITRETN (30),ILOCN (30),KLOCN (54)
2   ,NSN (12,20)
DOUBLE PRECISION PNEW
C   COMMON SAVE(12),B,DD,T,A,D,PD
DOUBLE PRECISION A(3,3),B(6),D(30),DD(6),PD(6,6),T(3,3)
C   DIMENSION BN(9)
DATA BN/2H N,2H V,2H W,2HND,2HVD,2HWD
1   ,6H ,6HODARBO,6HOTANGT/
C
600 FORMAT(1H0,51X,17HCOVARIANCE MATRIX)
601 FORMAT(1H0,37X,44HTHE FOLLOWING STATES HAVE NEGATIVE DIAGONALS//47
1X,5HSTATE,10X,8HVARIANCE)
602 FORMAT(41X,A6,1X,A3,D22.8)
603 FORMAT(1H0,7X,6(12X,A6))
604 FORMAT(A6,6(15X,A3))
605 FORMAT(8X,3HSTD,1X,6D18.8)
606 FORMAT(1X,A6,1X,A3,4X,F12.9,5(6X,F12.9))
607 FORMAT(1H )
C
C   OUTPUT WILL APPEAR UNCONDITIONALLY IN COORDINATES OF ESTIMATE
C   IF LD.NE.0 OUTPUT WILL FIRST APPEAR IN ALTERNATE COORDINATES
C   IF SPACECRAFT STATE ELEMENTS ARE PRESENT
C   LD=1, OUTPUT IN DARBOUTX
C   LD=2, OUTPUT IN LOCAL TANGENT PLANE
C
WRITE(6,600)
NP=NNP
KP=1
DO 2 I=1,NP
IF(PNEW(I,1).GT.0.) GO TO 2
IF(KP.EQ.0) GO TO 1
WRITE(6,601)
KP=0
1 WRITE(6,602) TRAKER(I),STNAMN(I),PNEW(I,I)
2 D(I)=DSQRT(DABS(PNEW(I,I)))
IF(KLOCN(I).NE.1) GO TO 20
IF(LD.EQ.0) GO TO 20
C   OUTPUT IN ALTERNATE COORDINATES
CALL DTRDB(X,X(4),T,LD)
DO 12 I=1,3
DO 12 J=1,3
12 A(I,J)=T(J,I)
CALL DTRANP(PNEW,30,A,A,PD,0)
DO 13 I=1,6
DD(I)=DSQRT(DABS(PD(I,I)))
13 PD(I,I)=1.
DO 14 I=2,6
DO 14 J=1,6
14 PD(J,I-1)=PD(J,I-1)/D(J)/D(I-1)
WRITE(6,604) BN(LD+7),(BN(I),I=1,6)
WRITE(6,605) (DD(I),I=1,6)
WRITE(6,607)
DO 15 I=1,6
15 WRITE(6,606) TRAKER(I),BN(I),(PD(I,J),J=1,1)
IF(NP.EQ.6) GO TO 20
DO 19 I=7,NP
CALL DMVTRN(A,PNEW(1,I),1,2)
DO 18 J=1,6
18 B(J)=B(J)/D(I)/D(J)
19 WRITE(6,606) TRAKER(I),STNAMN(I),(B(J),J=1,6)
20 CONTINUE
C   COORDINATES OF ESTIMATE
K=0
21 IF(NP.LE.K) GO TO 99
KV=NP-K
JJ=6
IF(KV.LT.6) JJ=KV

```

```

COVT0001
COVT0002
COVT0003
COVT0004
COVT0005
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COVT0065
COVT0066
COVT0067
COVT0068
COVT0069
COVT0070
COVT0071
COVT0072
COVT0073
COVT0074

```

```

DO 23 I=1, JJ
L=K+I
DO 22 J=1, I
M=K+J
22 PD(I, J)=PNEW(L, M)/D(L)/D(M)
23 PD(I, I)=1.
KV=L-JJ+1
WRITE(6, 603) (TRAKER(I), I=KV, L)
WRITE(6, 604) BN(7), (STNAMN(I), I=KV, L)
WRITE(6, 605) (D(I), I=KV, L)
WRITE(6, 607)
KV=KV-1
DO 24 I=1, JJ
KP=KV+I
24 WRITE(6, 606) TRAKER(KP), STNAMN(KP), (PD(I, J), J=1, I)
IF(KP.EQ.NP) GO TO 99
KP=KP+1
DO 26 I=KP, NP
DO 25 J=1, JJ
L=KV+J
25 B(J)=PNEW(L, I)/D(L)/D(I)
26 WRITE(6, 606) TRAKER(I), STNAMN(I), (B(J), J=1, JJ)
K=K+JJ
GO TO 21
99 RETURN
END

```

```

COVT0075
COVT0076
COVT0077
COVT0078
COVT0079
COVT0080
COVT0081
COVT0082
COVT0083
COVT0084
COVT0085
COVT0086
COVT0087
COVT0088
COVT0089
COVT0090
COVT0091
COVT0092
COVT0093
COVT0094
COVT0095
COVT0096
COVT0097
COVT0098
COVT0099

```

Subroutine: CRITA

Purpose: To output the time at which a body starts or stops occulting the vehicle.

Calling Sequence: CALL CRITA (TIME, BNAM, CNAM, NOUT, KSW)

Input and Output

I/φ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	TIME			Seconds	Time for output.
I	BNAM				Name of occulting body (BCD).
I	CNAM				Name of body being occulted.
I	NOUT				Output tape unit.
I	KSW				If KSN = 1, occulting just starting. KSN = 2, occulting just stopped.

Common storages used: None

Subroutines required: TIMES

CRITA-1

```
$IBFTC MC13CA NOREF,M94,NODD,XR3
CMC13CA CRITA OUTPUTS OCCULTING BODY
SUBROUTINE CRITA(TIME,BNAM,CNAM,NOUT,KSW)
DIMENSION ANAM(2), A(6)
DATA (ANAM(J),J=1,2)/6HSTARTS,6H STOPS/
1 CALL TIMES(TIME,D,A)
SW=ANAM(KSW)
WRITE (NOUT,100) A,BNAM,SW,CNAM
100 FORMAT(6A6,1X,1A6,1X,1A6,1X,22HOCCULTING VEHICLE FROM,1X,1A6)
RETURN
END
```

```
CRIA000
CRIA001
CRIA002
CRIA003
CRIA004
CRIA005
CRIA006
CRIA007
CRIA008
CRIA009
```

Subroutine: CRITIC

Purpose: Takes station-related or beacon-related critical events from the STIME or BTIME arrays and forms arrays of station on-off times or beacon on-off times - SSTART,SSTOP or BSTART,BSTOP.

Calling sequence: CALL CRITIC(NCX,IXC,XTIME,IXTIME,IXEE,KXMAX,
TIM,XSART,XSTOP,KXN,KAMAX)

Input and Output

I/∅	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	NCX				NCX = 10, beacons = 12, stations.
I	IXC	As needed			Array of station numbers, ISC or Array of beacon numbers, IBC.
I	XTIME	As needed			Array of critical event times from SBEV2.
I	IXTIME	As needed			Array of critical event types, corresponds to XTIME.
I	IXEE	As			ISEE or IBEE array; in view indication for stations/beacons listed in IXC array.
I	KXMAX				Number of critical events in XTIME array.
I	TIM	(2)			Time span over which critical events were found.
∅	XSTART	(75)			On times off times, and related station/beacon number.
∅	XSTOP	(75)			
∅	KXN	(75)			
∅	KAMAX				Number of on-off pairs in XSTART,XSTOP.

Common storages used: /INPCOM/

Subroutines required: BARN, SORDR2

CRITIC-1

Functions of CRITIC

CRITIC receives as inputs a list, IXTIM, of critical event types such as station/beacon into view, station/beacon out of view, occulting starts, and occulting stops; a list, XTIME, of times associated with these events; a list of station/beacon numbers, IXC; a corresponding list of in-view indicators, IXEE, that describe conditions at the time TIM(1), which is the start time for the list of critical events; and a stop time, TIM(2), which is the end time for the list of critical events. From these input data, CRITIC forms three corresponding lists: XSTART, a list of station/beacon on times, XSTOP, a list of station/beacon off times, and KXN, a list of station/beacon numbers. On times for which no corresponding off time is found are assigned off times of TIM(2). One extra on time for each station/beacon is added to the list; this on time is set very large and merely fulfills a program logic requirement.

To each on time is added a small, "random" number representing the acquisition delay after the station/beacon in-view event. The standard deviation for this delay is found in C(114) of INPCOM. Subroutine BARN is the random number generator.

The Differential Correction Program requires that data arcs not exceed a maximum time span. Therefore, CRITIC examines all on-off time pairs and appropriately breaks up, by inserting additional off and on times, any station/beacon on periods longer than XMAX, which is stored in C(113) of INPCOM.

Next, CRITIC eliminates on periods shorter than XMIN (found in C(112)).

Finally, CRITIC sorts all three output lists by ascending order of on times in list XSTART.

CRITIC-2

```

$IBFTC MC13CC NOREF,M94,NODD,XR3
CMC13CC CRITIC SORTS START-STOP ARRAYS FROM STIME OR BTIME ARRAY CRIC0001
SUBROUTINE CRITIC(NCX,IXC,XTIME,IXTIM,IXEE,KXMAX,TIM,XSTART,XSTOP, CRIC0002
1KXN,KAMAX) CRIC0003
C SUBROUTINES REQUIRED SORDR2 BARN CRIC0004
COMMON/INPCOM/C(700) CRIC0005
DIMENSION XTIME(1), IXTIM(1), IXC(1), IXEE(1) CRIC0006
1, KXN(1), XSTART(1), XSTOP(1) CRIC0007
2, TIM(2) CRIC0008
EQUIVALENCE (C(112),XMIN), (C(113),XMAX), (C(114),RDELAY) CRIC0009
LOGICAL NBEAC,STOP CRIC0010
DATA C2XR3/ 3.46410162/ CRIC0011
DELA=C2XR3*RDELAY CRIC0012
NBEAC=NCX.EQ.10 CRIC0013
TIMEND=XTIME(KXMAX) CRIC0014
STOP=TIMEND.NE.TIM(2)+99999. CRIC0015
IF(STOP) GO TO 50 CRIC0016
TIMEND=TIM(2) CRIC0017
KXMAX=KXMAX-1 CRIC0018
50 CONTINUE CRIC0019
KA=1 CRIC0020
KM=0 CRIC0021
IF(NBEAC) GO TO 1 CRIC0022
IF(IXTIM.EQ.13.AND.XTIME.EQ.TIM) GO TO 19 CRIC0023
1 DO 5 I=1,NCX CRIC0024
K=IXC(I) CRIC0025
IF(K.EQ.0) GO TO 6 CRIC0026
IF(IXEE(I)) 5,2,2 CRIC0027
2 KXN(KA)=I CRIC0028
XSTART(KA)=XTIME(KM) CRIC0029
IF(KM.EQ.0) XSTART(KA)=TIM CRIC0030
IF(KM.EQ.KXMAX) GO TO 35 CRIC0031
KMP1=KM+1 CRIC0032
DO 3 L=KMP1,KXMAX CRIC0033
IF(IXTIM(L).EQ.13) GO TO 4 CRIC0034
IF(IXTIM(L)+I) 3,4,3 CRIC0035
3 CONTINUE CRIC0036
35 CONTINUE CRIC0037
XSTOP(KA)=TIMEND CRIC0038
GO TO 41 CRIC0039
4 XSTOP(KA)=XTIME(L) CRIC0040
41 KA=KA+1 CRIC0041
5 CONTINUE CRIC0042
6 CONTINUE CRIC0043
IF(KM.EQ.KXMAX) GO TO 100 CRIC0044
KM=KM+1 CRIC0045
K=IXTIM(KM) CRIC0046
IF(K.EQ.13) GO TO 20 CRIC0047
IF(K) 17,1000,8 CRIC0048
8 IXEE(K)=0 CRIC0049
XSTART(KA)=XTIME(KM) CRIC0050
KXN(KA)=K CRIC0051
IF(KM.EQ.KXMAX) GO TO 70 CRIC0052
KM1=KM+1 CRIC0053
DO 10 L=KM1,KXMAX CRIC0054
IF(IXTIM(L) - 13) 9,12,9 CRIC0055
IF(IXTIM(L) + K) 10,12,10 CRIC0056
9 CONTINUE CRIC0057
70 XSTOP(KA)=TIMEND CRIC0058
11 KA=KA+1 CRIC0059
GO TO 6 CRIC0060
12 XSTOP(KA)=XTIME(L) CRIC0061
GO TO 11 CRIC0062
17 JK=-K CRIC0063
IXEE(JK)=-1 CRIC0064
GO TO 6 CRIC0065
19 KM=1 CRIC0066
20 CONTINUE CRIC0067
IF(KM.EQ.KXMAX) GO TO 100 CRIC0068
KM=KM+1 CRIC0069
K=IXTIM(KM) CRIC0070
IF(K.EQ.14) GO TO 1 CRIC0071
IF(K) 21,1000,22 CRIC0072
21 JK=-K CRIC0073
IXEE(JK)=-1 CRIC0074

```

	GO TO 20	CRIC0075
22	IXEE(K)=0	CRIC0076
	GO TO 20	CRIC0077
100	KAMAX=KA-1	CRIC0078
	KDLET=0	CRIC0079
	DO 110 I=1,KAMAX	CRIC0080
110	XSTART(I)=XSTART(I)+DELA* BARN(1)	CRIC0081
	KA=KAMAX	CRIC0082
	DO 120 I=1,KA	CRIC0083
	IF(XSTOP(I)-XSTART(I)-XMIN) 101,102,102	CRIC0084
101	XSTART(I)=TIM(2)+9999.	CRIC0085
	KDLET=KDLET+1	CRIC0086
	GO TO 120	CRIC0087
102	IF(XSTOP(I)-XSTART(I)-XMAX) 120,120,103	CRIC0088
103	XSTART(KAMAX+1)=XSTART(I)+XMAX	CRIC0089
	XSTOP(KAMAX+1) = XSTOP(I)	CRIC0090
	XSTOP(I)=XSTART(KAMAX+1)	CRIC0091
	KAMAX=KAMAX+1	CRIC0092
	KXN(KAMAX)=KXN(I)	CRIC0093
105	IF(XSTOP(KAMAX)-XSTART(KAMAX)-XMAX) 119,120,104	CRIC0094
104	XSTART(KAMAX+1)=XSTART(KAMAX)+XMAX	CRIC0095
	XSTOP(KAMAX+1)=XSTOP(KAMAX)	CRIC0096
	XSTOP(KAMAX)= XSTART(KAMAX+1)	CRIC0097
	KAMAX=KAMAX+1	CRIC0098
	KXN(KAMAX)=KXN(I)	CRIC0099
	GO TO 105	CRIC0100
119	IF(XSTOP(KAMAX)-XSTART(KAMAX)-XMIN.LT.0.) KAMAX=KAMAX-1	CRIC0101
120	CONTINUE	CRIC0102
	CALL SORDR2(XSTART,KXN,KAMAX,2,XSTOP)	CRIC0103
	KAMAX=KAMAX-KDLET	CRIC0104
	IF(STOP) GO TO 200	CRIC0105
201	KAMAX=KAMAX+1	CRIC0106
	XSTART(KAMAX)=XTIME(KXMAX+1)	CRIC0107
	GO TO 200	CRIC0108
1000	WRITE (6,700)	CRIC0109
700	FORMAT(1H0,20H****CRITIC ERROR****)	CRIC0110
200	RETURN	CRIC0111
	END	CRIC0112

Subroutine: CRITØ

Purpose: To output at a station (or beacon) critical event.

Calling Sequence: CALL CRITØ (TIME,SNAME,K,R,EL,NØUT,NHEAD)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	TIME			Seconds	Time of critical event for output.
I	SNAME				Station name.
I	K				If K = 1, station is turned off K = 2, station is turned on
I	R	(3)			Vehicle position with respect to station.
I	EL				Vehicle elevation.
I	NØUT				Output tape number
I/Ø	NHEAD				If NHEAD = 1, a header is output and NHEAD is set = 2. NHEAD = 2, no header is output.

Common storages used: None

Subroutines required: FNØRM,ARKTNS,TIMES

CRITØ-1

\$IBFTC MC13CR	NOREF,M94,NODD,XR3	
CMC13CR	CRITO	CRI00010
	SUBROUTINE CRITO(TIME,SNAME,K,R,EL,NOUT,NHEAD)	CRI00020
	DIMENSION R(3), A(6), SWCH(2)	CRI00030
	DATA RTD,(SWCH(J),J=1,2)/57.2957796,4H OFF,3H ON/	CRI00040
C	SUBROUTINES REQUIRED ARE FNORM,ARKTNS,TIMES	CRI00050
	SW=SWCH(K)	CRI00060
	GO TO (1,2),NHEAD	CRI00070
1	CONTINUE	CRI00080
	NHEAD=2	CRI00090
	WRITE (NOUT,100)	CRI00100
100	FORMAT(1H0,30X,37HSTATION CRITICAL EVENT AND CONDITIONS/1H ,9X,15HCRI00110	
	1TIME FROM EPOCH,20X,5HEVENT,17X,5HRANGE,10X,7HAZIMUTH,6X,9HELEVATICRI00120	
	2ON)	CRI00130
2	CONTINUE	CRI00140
	RM=FNORM(R)	CRI00150
	AZ=ARKTNS(180,R(1),R(2))	CRI00160
	AZ=AZ*RTD	CRI00170
	ELD=EL*RTD	CRI00180
	CALL TIMES(TIME,D,A)	CRI00190
	WRITE (NOUT,101) A,SNAME,SW,RM,AZ,ELD	CRI00200
101	FORMAT(6A6,7X,2A6,8X,E15.8,3X,F8.3,6X,F7.3)	CRI00210
	RETURN	CRI00220
	END	CRI00230

Subroutine: CRØSIM

Purpose: Solves simultaneous linear algebraic equations in
double precision.

Calling Sequence: CALL CRØSIM

Common storages used: /CRØCOM/

Subroutines required: None.

CRØSIM-1

1. Use of CRØSIM

Common assignments, CCRØ(86)

Location	Name	Dimension	Description
1	A	d(6,7)	Input: Augmented matrix of coefficients. Output: Solution vector will appear in the first L elements of column L+1.
85	L		Number of rows of A to be considered.
86	ISING		Output singularity indicator. ISING = 0: Solution found ISING = 1: Singular matrix, no solution found.

CRØSIM will solve up to six linear equations. More equations may be accommodated by altering the dimensions of the input/output matrix A and making the appropriate changes in the common size and assignments. The upper left L by L subset of A must be loaded with the equation coefficients; the first L elements of column L+1 must be the constant terms. The original matrix is destroyed during computation. The solution is developed in the first L elements of column L+1. All elements of A and all internal computations are in double precision.

CRØSIM-2

2. Method of Solution

Reference: Hildebrand, F.B.: Methods of Applied Mathematics, Prentice-Hall, Inc., Englewood Cliffs, N.J., 1952, p. 503, et seq.

The method of Crout, described in the reference, has been modified slightly to improve accuracy. The modification is a re-ordering of the equations as necessary to use the available diagonal element of greatest magnitude as a divisor in each step of the reduction. If, at any step, the diagonal element of greatest magnitude is identically zero, the singularity flag ISING is set to 1 and the reduction is abandoned. If no singularity condition occurs, ISING is set to zero.

Crout's method for systems with real coefficients proceeds from the augmented matrix of the system

$$M = \left(\begin{array}{cccc|c} a_{11} & a_{12} & \dots & a_{1n} & c_1 \\ a_{21} & a_{22} & \dots & a_{2n} & c_2 \\ \vdots & \vdots & & \vdots & \vdots \\ \vdots & \vdots & & \vdots & \vdots \\ a_{n1} & a_{n2} & \dots & a_{nn} & c_n \end{array} \right) = (a \mid c),$$

which may be considered as partitioned into the coefficient matrix a and the column vector c , to an auxiliary matrix

$$M' = \left(\begin{array}{cccc|c} a'_{11} & a'_{12} & \dots & a'_{1n} & c'_1 \\ \vdots & \vdots & & \vdots & \vdots \\ a'_{21} & a'_{22} & \dots & a'_{2n} & c'_2 \\ \vdots & \vdots & & \vdots & \vdots \\ \vdots & \vdots & & \vdots & \vdots \\ a'_{n1} & a'_{n2} & \dots & a'_{nn} & c'_n \end{array} \right) = (a' \mid c')$$

CRØSIM-3

of the same dimensions, and thence to the required solution vector

$$x = \begin{pmatrix} x_1 \\ x_2 \\ \cdot \\ \cdot \\ x_n \end{pmatrix}$$

The procedure for finding the elements of M' from those of M is described by the following rules:

1. The elements of M' are determined in the following order: elements of the first column, then elements of the first row to the right of the first column; elements of the second column below the first row, then elements of the second row to the right of the second column; and so on until all elements are determined.

2. The column operations may be divided into three steps as follows:

a. Each element a'_{ij} on or below the principal diagonal of M' is obtained by subtracting from the corresponding a_{ij} of M the sum of the products of elements in the i^{th} row and corresponding elements in the j^{th} column of M' , all uncalculated elements being imagined to be zeros. That is

$$a'_{ij} = a_{ij} - \sum_{k=1}^{j-1} a'_{ik} a'_{kj} \quad (i \geq j) .$$

For the first column, this is clearly no operation at all.

b. The magnitude of the diagonal element a'_{ii} of the i^{th} column of M' is compared with the magnitudes of the elements of the i^{th} column below the diagonal a'_{ik} , $k = i+1, n$. If the largest of these, say a'_{ik} , is greater in magnitude than a'_{ii} , then the entire i^{th} and k^{th} rows of M (and M') are interchanged. This step is purely a numerical device to minimize round-off errors.

CROSSIM-4

c. Each element a'_{ij} below the principal diagonal is finally determined by division by the diagonal element a'_{jj} of M' . That is

$$a'_{ij} = a'_{ij} / a'_{jj} \quad (i > j)$$

3. Row operations consist of a single step: Each element a'_{ij} to the right of the principal diagonal is obtained by subtracting from the corresponding element a_{ij} of M the sum of the products of elements in the i^{th} row and corresponding elements of the j^{th} column of M' , all calculated elements being imagined to be zeros. That is

$$a'_{ij} = a_{ij} - \sum_{k=1}^{i-1} a'_{ik} a'_{kj} \quad (i < j)$$

Clearly for the first row this involves no operation at all.

The procedure for obtaining the final solution vector from the matrix a' and the vector c' into which M' is partitioned is described by the following rules:

1. The elements of x are determined in the reverse order x_n, x_{n-1}, \dots, x_1 from the last element to the first.

2. The last element x_n is determined by dividing the last element c'_n of c' by the last element a'_{nn} on the principal diagonal of M' . That is

$$x_n = c'_n / a'_{nn}$$

3. Each of the remaining elements x_i of x is determined by subtracting from the corresponding element c'_i of c' the sum of the products of elements in the i^{th} row of a' by corresponding elements of the column x , followed by a division by the diagonal element a'_{ii} of M' , all uncalculated elements of x being imagined to be zeros. That is

CR0SIM-5

$$x_i = \frac{c_i' - \sum_{k=i+1}^n a_{ik}' x_k}{a_{ii}'}$$

Core storage space is saved in CRØSIM by causing the matrices M and M' to share the same storage locations.

CRØSIM-6

\$IBFTC MCL34Q M94,NODD,XR3	
CMC134Q DOUBLE PRECISION SIMULTANEOUS LINEAR EQUATION SOLVER	CRSM0001
SUBROUTINE CROSIM	CRSM0002
COMMON /CROCOM/CCRO(86)	CRSM0003
DOUBLE PRECISION A(6,7)	CRSM0004
EQUIVALENCE (CCRO(1),A), (CCRO(85),L), (CCRO(86),ISING)	CRSM0005
C	CRSM0006
DOUBLE PRECISION SUM	CRSM0007
C	CRSM0008
ISING=0	CRSM0009
M=L	CRSM0010
N=M+1	CRSM0011
IK=1	CRSM0012
1 IL=IK	CRSM0013
SUM=DABS(A(IK,IK))	CRSM0014
DO 2 I=IK,M	CRSM0015
IF(SUM.GE.DABS(A(I,IK))) GO TO 2	CRSM0016
IL=I	CRSM0017
SUM=DABS(A(I,IK))	CRSM0018
2 CONTINUE	CRSM0019
IF(IK.EQ.IL) GO TO 4	CRSM0020
DO 3 J=1,N	CRSM0021
SUM=-A(IK,J)	CRSM0022
A(IK,J)=A(IL,J)	CRSM0023
3 A(IL,J)=SUM	CRSM0024
4 IL=IK+1	CRSM0025
IF(A(IK,IK).EQ.0.) GO TO 11	CRSM0026
DO 5 I=IL,M	CRSM0027
5 A(I,IK)=A(I,IK)/A(IK,IK)	CRSM0028
6 JJ=IK-1	CRSM0029
IL=IK+1	CRSM0030
IF(JJ.EQ.0) GO TO 8	CRSM0031
DO 7 J=IL,N	CRSM0032
DO 7 I=1,JJ	CRSM0033
7 A(IK,J)=A(IK,J)-A(IK,I)*A(I,J)	CRSM0034
IF(IK.EQ.M) GO TO 10	CRSM0035
8 JJ=IK	CRSM0036
IK=IK+1	CRSM0037
DO 9 I=IK,M	CRSM0038
DO 9 J=1,JJ	CRSM0039
9 A(I,IK)=A(I,IK)-A(I,J)*A(J,IK)	CRSM0040
IF(IK=M) 1,6,1	CRSM0041
10 IF(A(M,M).NE.0.) GO TO 12	CRSM0042
11 ISING=1	CRSM0043
GO TO 14	CRSM0044
12 DO 13 I=1,M	CRSM0045
JJ=M-I	CRSM0046
IL=JJ+1	CRSM0047
A(IL,N)=A(IL,N)/A(IL,IL)	CRSM0048
IF(JJ.EQ.0) GO TO 14	CRSM0049
DO 13 J=1,JJ	CRSM0050
13 A(J,N)=A(J,N)-A(IL,N)*A(J,IL)	CRSM0051
14 RETURN	CRSM0052
END	CRSM0053

Subroutine: CRØSS

Purpose: Computes the vector cross product of two vectors.

Calling Sequence: CALL CRØSS (A,B,C)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	A,B	(3)			Input vectors
Ø	C	(3)			A X B

Common storages used: None

Subroutines required: None

CRØSS-1

```
$IBFTC MC13MK XR3,M94,NODD,LIST
  SUBROUTINE CROSS (A,B,C)
  DIMENSION A(3),B(3),C(3),IN(4)
  DATA      IN/2,3,1,2/
  DO 1 I=1,3
  J = IN(I)
  K = IN(I+1)
1 C(I) = A(J)*B(K)-A(K)*B(J)
  RETURN
  END
```

```
CROS0001
CROS0002
CROS0003
CROS0004
CROS0005
CROS0006
CROS0007
CROS0008
```

Subroutine: DATINP

Purpose: To convert from calendar date (in date format)
 to double-precision seconds from 0^h, Jan 1, 1950.

Calling Sequence: CALL DATINP(DI,DPSEC)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	DI	3			DI(1) = YR MO. } Date DI(2) = DAY HR MIN. } For- DI(3) = SEC. } mat
Ø	DPSEC	d			Double-precision seconds from 0 ^h , Jan 1, 1950.

Common storages used: 8 cells.

Subroutines required: None

DATINP-1

Date Format:

The calendar date is represented by a three-element, single-precision array DI.

Let:

YR be the year number from 1900;
i.e., the year 1967 is represented as
67. ; the year 2001 as 101.

MO be the number of the month.

DAY be the day of the month.

HR be the whole hour on the 24-hour clock.

MIN be the whole minutes past the hour.

SEC be the whole and fractional seconds.

Then:

$$DI(1) = YR. \times 10^2 + MO.$$

$$DI(2) = DAY \times 10^4 + HR. \times 10^2 + MIN.$$

$$DI(3) = SEC.$$

Example:

JAN 24, 1967, 11^h 26^m 16.482^s is represented as:

$$DI(1) = 6701.$$

$$DI(2) = 241126.$$

$$DI(3) = 16.482$$

DATINP-2

```

SIBFTC MC13TV XR3,M94,NODD,LIST
SUBROUTINE DATINP(DI,DPSEC)
C
C      DI(1) = YEAR MONTH.                (INPUT)
C      DI(2) = DAY HR MIN.                (INPUT)
C      DI(3) = SECONDS.                    (INPUT)
C      DPSEC = DOUBLE PRECISION SECONDS FROM JAN 0, 1950 (OUTPUT)
C
      DOUBLE PRECISION DPSEC
      COMMON          Y,I(5),JJ
      DIMENSION      J(11),DI(3)
      DATA          J/31,59,90,120,151,181,212,243,273,304,334/
C
      I=DI(1)
      I(2)=DI(2)
      IF(I.LT.5000.OR.I(2).LT.0) GO TO 2
      I(3)=I(2)/10000
      I(4)=I(2)-10000*I(3)
      I(2)=I(4)/100
      I(5)=I(4)-100*I(2)
      Y=60*(I(5)+60*I(2))
      DPSEC=Y+DI(3)
      GO TO 10
2 WRITE(6,700) DI
700 FORMAT(22H1DATINP EXIT, YRMO. =,F10.2,12H DAYHRMIN. =,F12.2,
16H SEC =,F9.6)
      STOP
10 I(3)=I(3)-1
      I(4)=I(1)/100
      I(1)=I(1)-100*I(4)-1
      IF(I.GT.11) GO TO 2
      IF(I) 2,13,12
12 JJ=I
      I(3)=I(3)+J(JJ)
13 I(5)=I(4)-4*(I(4)/4)
      IF(I(1)/2+I(5).LE.0) I(3)=I(3)-1
      Y=I(3)+(2+1461*(I(4)-50))/4
      DPSEC=DPSEC+86400.*Y
      RETURN
      END

```

```

DATI0001
DATI0002
DATI0003
DATI0004
DATI0005
DATI0006
DATI0007
DATI0008
DATI0009
DATI0010
DATI0011
DATI0012
DATI0013
DATI0014
DATI0015
DATI0016
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DATI0018
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DATI0021
DATI0022
DATI0023
DATI0024
DATI0025
DATI0026
DATI0027
DATI0028
DATI0029
DATI0030
DATI0031
DATI0032
DATI0033
DATI0034
DATI0035
DATI0036
DATI0037
DATI0038

```

Subroutine: DATØUP

Purpose: To convert from double-precision seconds from
0^h Jan 1, 1950 to calendar date.

Calling Sequence: CALL DATØUP(DPSEC, D, K)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	DPSEC	d			Double-precision seconds from 0 ^h Jan 1, 1950
Ø	D	3			D(1) = YR MO. } Date D(2) = DAY HR MIN } For - D(3) = SEC. } mat*
I	K				K = 0 causes the calendar and Julian dates to be written on the system output tape.

* See DATINP Description.

Common storages used: 10 cells.

Subroutines required: None.

DATØUP-1

```

$IBFTC MC13TW XR3,M94,NODD,LIST
SUBROUTINE DATOUP(DPSEC,D,K)
C
C DPSEC = DOUBLE PRECISION SECONDS FROM 1950 (INPUT)
C D(1) = YR MO. FORMAT OF DATINP (OUTPUT)
C D(2) = DAY HR MIN. FORMAT OF DATINP (OUTPUT)
C D(3) = SEC. FORMAT OF DATINP (OUTPUT)
C K= 0 GIVES CALENDAR DATE FORMAT OUTPUT
C
DOUBLE PRECISION DPSEC, TDUM
DIMENSION C(12), D(3), IY(4)
COMMON Y(8), TDUM
EQUIVALENCE (Y(2),IY)
C
DATA C/ 3HJAN, 3HFEB, 3HMAR, 3HAPR, 3HMAY, 3HJUN
1, 3HJUL, 3HAUG, 3HSEP, 3HOCT, 3HNOV, 3HDEC /
C
601 FORMAT(1H ,A3,I3,1H,,I5,1H,,I3,5H HRS.,I3,5H MIN.,F9.5,4H SEC,48X,
111HJULIAN DATE,F10.0,F8.8)
602 FORMAT(21HONEG DAYS IN DATOUP =,F8.0)
C
TDUM=DPSEC
Y(7)=AINT(TDUM/86400.D0)
IF(TDUM.GT.0.)GO TO 10
WRITE(6,602) Y(7)
GO TO 999
10 CONTINUE
TDUM=TDUM-Y(7)*86400.
Y(8)=TDUM/86400.D0
IY(1) = Y(7)
IY(2) = IY(1)/365
11 IY(3) = IY(1)-(1461*IY(2)+1)/4
IF (IY(3).GE.0) GO TO 12
IY(2) = IY(2)-1
GO TO 11
12 IY(4) = IY(2)-2-4*(IY(2)/4)
IY(2) = IY(2)+50
JJ = 0
KD = 0
13 CONTINUE
MD = KD
JJ = JJ+1
GO TO (14,16,14,15,14,15,14,14,15,14,15,14) ,JJ
14 KD = KD+31
GO TO 17
15 KD = KD+30
GO TO 17
16 IF (IY(4).EQ.0) KD=KD+1
KD = KD+28
17 IF (KD.LE.IY(3)) GO TO 13
20 IY(1) = IY(3)-MD+1
Y(1) = C(JJ)
D(1)=JJ+100*IY(2)
IY(2) = IY(2)+1900
C
30 CONTINUE
Y(4)=AINT(TDUM/3600.D0)
TDUM=TDUM-Y(4)*3600.
Y(5)=AINT(TDUM/60.D0)
TDUM=TDUM-Y(5)*60.
IY(3)=Y(4)
IY(4)=Y(5)
Y(6)=TDUM
YY=IY(4)+100*(IY(3)+100*IY(1))
D(2)=YY
D(3)=Y(6)
C
40 CONTINUE
Y(8) = Y(8)+0.5
YY = AINT(Y(8))
Y(8) = Y(8)-YY
Y(7) = Y(7)+YY+2433282.
IF (K.NE.0) GO TO 999
WRITE (6,601) Y
999 RETURN
END

```

DAT00001
 DAT00002
 DAT00003
 DAT00004
 DAT00005
 DAT00006
 DAT00007
 DAT00008
 DAT00009
 DAT00010
 DAT00011
 DAT00012
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 DAT00014
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 DAT00016
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 DAT00019
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 DAT00073
 DAT00074

Subroutine: DATØUT

Purpose: Converts whole and fractional days since 1950 Jan 0.0 to date format. Optionally writes out calendar and Julian dates.

Calling Sequence: CALL DATØUT (TW,TF,YW,YF,K)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	TW			days	Whole days since 1950 Jan 0.0
I	TF			days	Fractional days
Ø	YW				Calendar date
Ø	YF				Time of day
I	K				If K = 0, date is written out If K ≠ 0, date is not written

Common storages used: 8 cells

Subroutines required: ERRØUT, TFRAC

DATØUT-1

Restrictions:

The permissible time range is from 1950 Jan 0.0 to but not including 2100 Jan 0.0. Times outside this range cause exit through ERRØUT.

Date Format:

The output quantities YW, YF are the floating point numbers

YW = YYMM.DD

YF = HHNN.SSSSS

where

YY = Year - 1900

MM = Month (number)

DD = Day of the month

HH = Hours since midnight

NN = Minutes

SSSSS = Seconds x 10³

DATAØUT-2

SIBFTC MC13DA XR3,M94,NODD,LIST	
SUBROUTINE DATOUT (TW,TF,YW,YF,K)	DAT00001
COMMON Y(8)	DAT00002
DIMENSION C(12),F(10),IY(4)	DAT00003
EQUIVALENCE(Y(2),IY)	DAT00004
DATA C(1)/36H JAN FEB MAR APR MAY JUN	DAT00005
1 /,C(7)/36H JUL AUG SEP OCT NOV DEC	DAT00006
2 /,F(1)/30H(6X,20HDATE 1950 OR BEFORE.)	DAT00007
3 /,F(6)/30H(6X,19HDATE 2100 OR LATER.) /	DAT00008
601 FORMAT(A6,I3,1H,,I5,1H,,I3,5H HRS,,I3,5H MIN,,F7.3,4H SEC,48X,	DAT00009
11HJULIAN DATE,F10.0,F8.8)	DAT00010
C	DAT00011
1 CALL TFRAC (TW,TF,Y(7),Y(8))	DAT00012
JJ = 1	DAT00013
IF (Y(7).LT.0.) GO TO 3	DAT00014
IF (Y(7).LT.54787.) GO TO 10	DAT00015
2 JJ = 6	DAT00016
3 CALL ERROUT (1,F(JJ))	DAT00017
C	DAT00018
10 CONTINUE	DAT00019
IY(1) = Y(7)	DAT00020
IY(2) = IY(1)/365	DAT00021
11 IY(3) = IY(1)-((1461*IY(2)+1)/4)	DAT00022
IF (IY(3).GE.0) GO TO 12	DAT00023
IY(2) = IY(2)-1	DAT00024
GO TO 11	DAT00025
12 IY(4) = IY(2)-2-4*(IY(2)/4)	DAT00026
IY(2) = IY(2)+50	DAT00027
JJ = 0	DAT00028
KD = 0	DAT00029
13 CONTINUE	DAT00030
MD = KD	DAT00031
JJ = JJ+1	DAT00032
GO TO (14,16,14,15,14,15,14,14,15,14,15,14) ,JJ	DAT00033
14 KD = KD+31	DAT00034
GO TO 17	DAT00035
15 KD = KD+30	DAT00036
GO TO 17	DAT00037
16 IF (IY(4).EQ.0) KD=KD+1	DAT00038
KD = KD+28	DAT00039
17 IF (KD.LE.IY(3)) GO TO 13	DAT00040
C	DAT00041
20 IY(1) = IY(3)-MD+1	DAT00042
Y(1) = C(JJ)	DAT00043
YW = FLOAT(IY(1)+100*(JJ+100*IY(2)))/100.	DAT00044
IY(2) = IY(2)+1900	DAT00045
C	DAT00046
30 Y(6) = Y(8)*24.	DAT00047
IY(3) = Y(6)	DAT00048
Y(6) = 60.*(Y(6)-FLOAT(IY(3)))	DAT00049
IY(4) = Y(6)	DAT00050
Y(6) = 60.*(Y(6)-FLOAT(IY(4)))	DAT00051
YY = IY(4)+100*IY(3)	DAT00052
YF = YY+Y(6)/100.	DAT00053
C	DAT00054
40 IF (K.NE.0) GO TO 999	DAT00055
Y(8) = Y(8)+0.5	DAT00056
YY = AINT(Y(8))	DAT00057
Y(8) = Y(8)-YY	DAT00058
Y(7) = Y(7)+YY+2433282.	DAT00059
WRITE (6,601) Y	DAT00060
999 RETURN	DAT00061
END	

Subroutine: DCP

Purpose: Main program for the Differential Correction Program.

Common storages used: //11 cells, /DCPCOM/, /EDTCOM/, /ESTCOM/

Subroutines required: DATINP, DIFCOR, MLESTT, RSIDUL, SETCAS, SETSTA, SETTAP

DCP-1

Discussion:

The Differential Correction Program was designed as a flexible post-flight data analysis program. It is capable of processing given data in any desired order, of generating an a priori estimate from the data itself, of propagating the state and covariance to a desired event, and of outputting residuals in a variety of forms.

The sequence of steps taken to process a given set of data is controlled at two levels. The case level is used for the definition of the quantities to be included in the state for the ensuing steps, and the process level is used to accomplish specific steps without redefinition of the state composition.

The main program described here controls the program flow as required to accomplish the specified tasks. The subroutines and secondary control routines used are listed below with functional descriptions (see the Flow Diagram below).

- | | |
|--------|--|
| SETTAP | Reads tape identification cards and checks or writes tape headers as required. Outputs tape identification cards. Sets IERR = 1 if any tape header disagrees with the corresponding identification card. |
| SETSTA | Locates the stations on the data tape, and loads working arrays for those stations from a larger array containing data for up to 50 stations. Computes station location vectors and transformations. Sets IERR = 2 if any station is not found in the array of available stations. |

DCP-2

SETCAS Reads option cards and sets internal option variables. Reads state definition cards and the associated overlay cards. Combines overlaid block data with a previously computed estimate from the estimation tape to form an a priori estimate of the state and its covariance. Sets internal indices used for interpretation of the state. Sets IERR for any invalid request in the state definition cards.

MLESTT Computes an estimate of the state from specified data for use as an a priori estimate, using maximum likelihood estimation in measurement space. Sets IERR = 15 if the MLE does not converge.

DIFCOR Controls the integration of vehicle equations of motion and variational equations for propagation or estimation of the vehicle state. Controls differential correction of the state. Sets IERR for any error noted.

RSIDUL Computes and outputs residuals of specified data from a specified state estimate.

Subroutines SETTAP and SETSTA comprise the program initialization for a given run, and are called only once during the run.

The DCP reads a case header card with the following contents:

Col 1-6 Number of estimate on the estimation tape to be used in building the a priori estimate.

Col 7-72 Hollerith case identification for user reference.

DCP-3

If the program encounters a blank card, or a card with a blank or 0 in 1-6, when it expects a case card, the run is terminated. The subroutine SETCAS performs case initialization, and is called once per case.

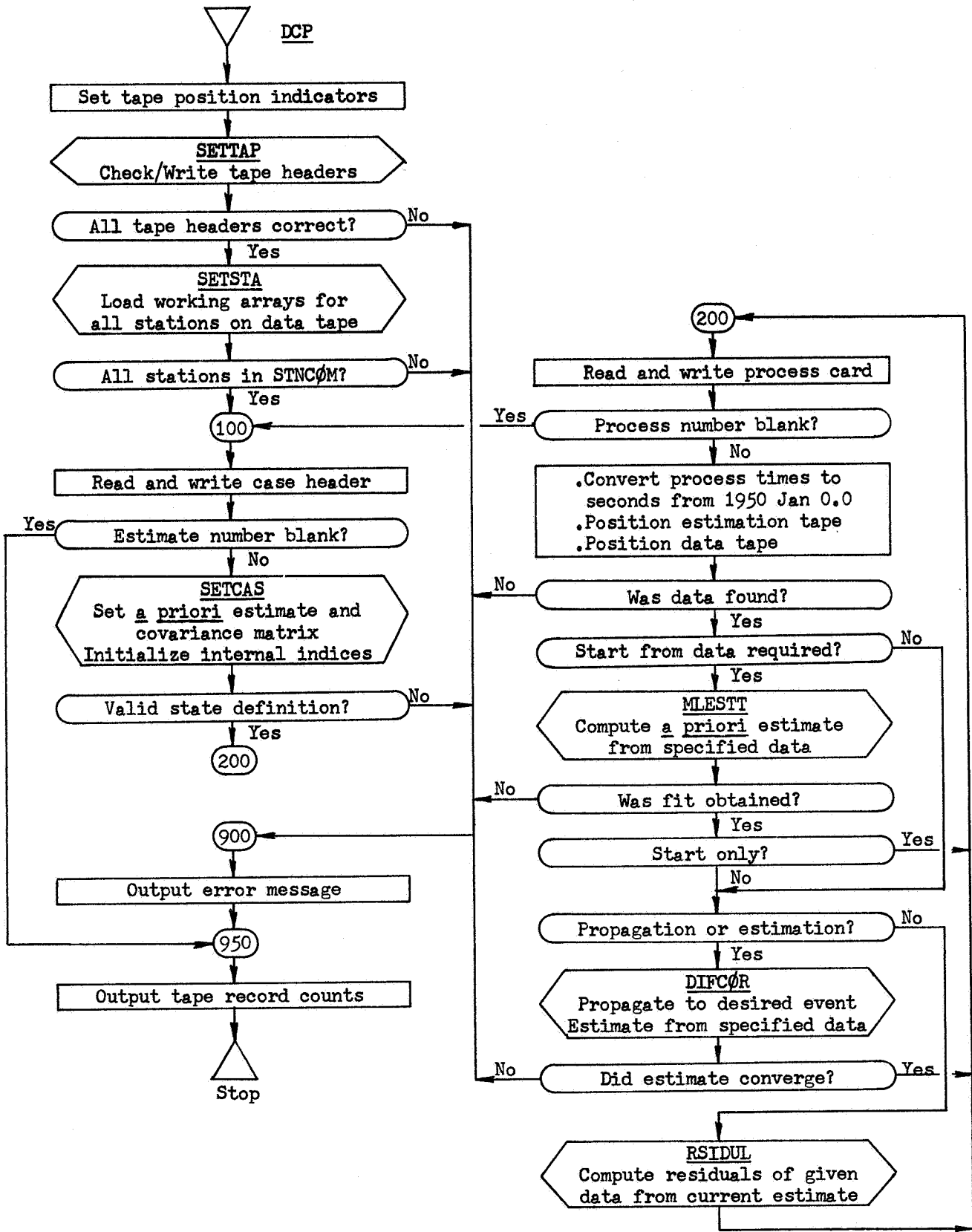
Following case initialization, the program reads a set of process control cards. For each, it searches the data tape for the data requested on the card and reads the first block of data into BUFDAT (EDTCOM). It then calls the appropriate data processing subroutine (MLESTT, DIFCOR,RSIDUL) and proceeds with the next process card. If the program encounters a card with a blank or 0 in column 1 when it expects a process card, the case is terminated and the next case card is read.

The process control card has the format

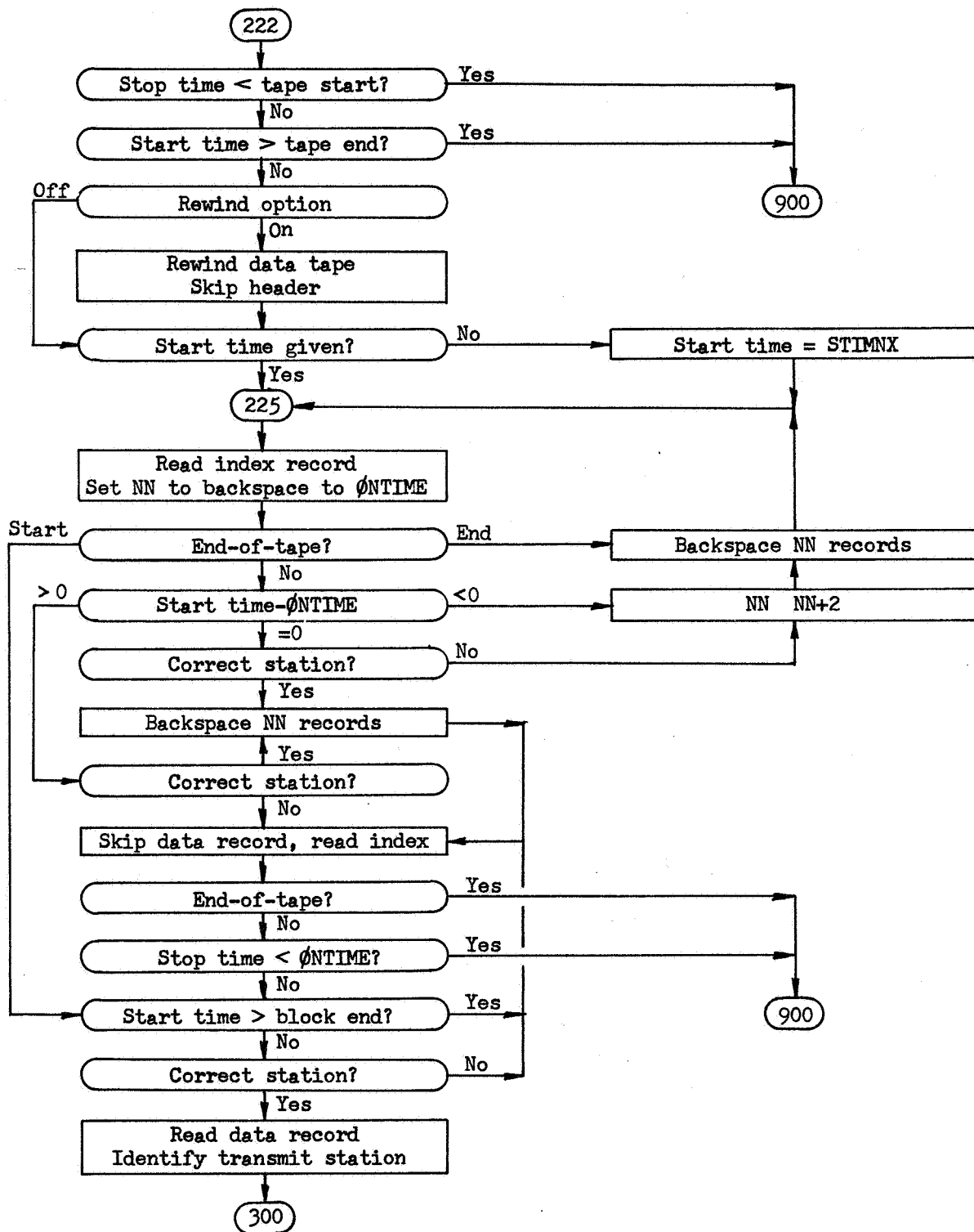
(2I1,A6,2(I3,I7,I2),I4,18I2)

Its contents are read into the array NPROC. The components of NPROC, their card locations, and their meanings are described in Reference 1.

DCP-4



DCP-5



DCP - Data Tape Search

DCP-6

```

SIBFTC MC13M3 XR3,M94,NODD,LIST
C   DCP - DIFFERENTIAL CORRECTION PROGRAM
C
C   DEVELOPED BY PHILCO-FORD CORPORATION
C   FOR GODDARD SPACE FLIGHT CENTER
C   CONTRACT NAS5-9939
C
C   REFERENCE TR-DA1508, PROGRAM DESCRIPTION AND THEORETICAL BASIS,
C   ORBIT DETERMINATION PROGRAM
C   TR-DA1509, SUBROUTINE DESCRIPTIONS AND LISTINGS,
C   ORBIT DETERMINATION PROGRAM
C   TR-DA1510, INPUT-OUTPUT SUMMARY,
C   ORBIT DETERMINATION PROGRAM
C
C   DECEMBER 1967
C
COMMON      /DCPCOM/CDCP(900)
DIMENSION   IPROC ( 5) ,NDATE(3,2) ,STIMNX(20)
1           ,NPROC(22) ,PDATE(3,2)
DOUBLE PRECISION  STIMR (2) ,TBEGIN      ,TFND
EQUIVALENC
1           ,(CDCP(111),IERR ) , (IPROC( 1),IPROCS) ,(NPROC( 1),NPROCS)
2           ,(CDCP(756),IPROC ) ,(IPROC( 2),ITRSTA) ,(NPROC( 2),NPROVR)
3           ,(CDCP(117),NESEND) ,(IPROC( 3),INDSTA) ,(NPROC( 3),NPRSTA)
4           ,(CDCP(118),NEST ) ,(IPROC( 4),IFTSTT) ,(NPROC( 4),NDATE )
5           ,(CDCP(731),NPROC ) ,(IPROC( 5),IFTSTP) ,(NPROC( 4),PDATE )
6           ,(CDCP(116),NRSFND) , (NPROC(10),NPREST)
7           ,(CDCP(123),STIMNX) , (NPROC(11),NREWND)
8           ,(CDCP(781),STIMR )
9           ,(CDCP(119),TBEGIN)
1          ,(CDCP(121),TFND )
C
COMMON      /ESTCOM/CEST(804)
DIMENSION   NAMSTA(20)
EQUIVALENC
1           ,(CEST( 7),NAMSTA) ,(CEST( 1),NFSPOS)
C
COMMON      /EDTCOM/INDDAT(40),BUFDAT(85,6)
DOUBLE PRECISION  ONTIME ,TLAST
EQUIVALENC
1           ,(INDDAT( 2),NEOT ) ,(INDDAT( 4),NTSTA )
2           ,(INDDAT( 5),NRCD ) ,(INDDAT(13),ONTIME)
C
COMMON      HEAD(11)
C
DIMENSION   DIAGN(132),NERR(19)
DATA  DIAGN( 1)/36HTAPE HEADER DOES NOT AGREE.
1 /,DIAGN( 6)/42HSTATION CANNOT BE FOUND IN BLOCK DATA.
2 /,DIAGN( 13)/42HSTATE REQUESTED FROM TAPE WITH NEGATIVE ES
2 /,DIAGN( 20)/18HTIMATE NUMBER.
3 /,DIAGN( 23)/36HSTATE TYPE NAMED DOES NOT EXIST.
4 /,DIAGN( 29)/24HSTATE TYPE DUPLICATED.
5 /,DIAGN( 33)/42HINCLUSION OF HARMONICS FOR TWO BODIES IS N
5 /,DIAGN( 40)/18HOT PERMITTED.
6 /,DIAGN( 43)/42HSTATE REQUESTED FROM TAPE ESTIMATE DOES NO
6 /,DIAGN( 50)/18HT EXIST THEREIN.
7 /,DIAGN( 53)/42HSTATION PARAMETERS REQUESTED FROM TAPE, ST
7 /,DIAGN( 60)/24HATION NOT IN ESTIMATE.
8 /,DIAGN( 64)/20HTOO MANY STATE VARIABLES.
9 /,DIAGN( 69)/42HTOO MANY EQUATION OF MOTION PARAMETERS.
DATA  DIAGN( 76)/42HCENTRAL BODY REQUESTED FROM TAPE ESTIMATE
1 /,DIAGN( 83)/30HDOES NOT AGREE WITH TAPE.
2 /,DIAGN( 88)/36HOBSRVING STATION NOT ON DATA TAPE.
3 /,DIAGN( 94)/42HPROCESS INTERVAL OUTSIDE DATA TAPE RANGE.
4 /,DIAGN(101)/42HPROCESS INTERVAL CONTAINS NO DATA FOR REQU
4 /,DIAGN(108)/18HSTED STATION.
5 /,DIAGN(111)/30HDATA START FAILED TO CONVERGE.
6 /,DIAGN(116)/36HTIME PRECEDES EPHEMERIS TAPE RANGE.
7 /,DIAGN(122)/36HTIME FOLLOWS EPHEMERIS TAPE RANGE.
8 /,DIAGN(128)/30HESTIMATION FAILED TO CONVERGE.
DATA  NERR/ 1, 6, 13, 23, 29, 33, 43, 53, 64, 69
1           , 76, 88, 94,101,111,116,122,128,133 /
C
5101 FORMAT(16,11A6)
5201 FORMAT(2I1,A6,2(I3,I7,I7),I4,18I2)

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```

6101 FORMAT(1H1,45X,27H*** CASE INITIALIZATION ***/9X,5HCASE ,11A6,15H,DCP 0075
1 TAPE ESTIMATE,I6,10H RFQUESTFD) DCP 0076
6201 FORMAT(1H1,48X,20H*** PROCESS CARD ***/103H0 PROC OVR STATION DCP 0077
1 START STOP NEST 1 2 3 4 5 6 7 8 DCP 0078
2 9 10 11 12/2I5,4X,A6,3X,2(I4,I7,I2),I6,12I4) DCP 0079
6901 FORMAT(15H0*** ERROR STOP/1X,11A6) DCP 0080
6902 FORMAT(20HOTAPE STATUS SUMMARY/26H RESIDUAL TAPE CONTAINS ,I3,13DCP 0081
1H RECORD PAIRS/26H ESTIMATION TAPE CONTAINS ,I3,13H RECORD PAIRS) DCP 0082
C DCP 0083
C**** RUN INITIALIZATION DCP 0084
C DCP 0085
C CHECK/WRITE TAPE HEADERS DCP 0086
1 CONTINUE DCP 0087
NFSP0S = 0 DCP 0088
NRSPOS = 0 DCP 0089
CALL SETTAP DCP 0090
IF (IERR,NE.0) GO TO 900 DCP 0091
C DCP 0092
C SET UP STATION DATA DCP 0093
2 CONTINUE DCP 0094
CALL SETSTA DCP 0095
IF (IERR,NF.0) GO TO 900 DCP 0096
C DCP 0097
C**** CASE INITIALIZATION DCP 0098
C DCP 0099
C READ CASE HEADER DCP 0100
100 CONTINUE DCP 0101
READ (5,5101) NEST,HEAD DCP 0102
IF (NEST,EQ.0) GO TO 950 DCP 0103
WRITE (6,6101) HEAD,NFST DCP 0104
C DCP 0105
C SET UP A PRIORI STATE AND COVARIANCE DCP 0106
CALL SETCAS DCP 0107
IF (IERR,NE.0) GO TO 900 DCP 0108
C DCP 0109
C**** PROCESS CONTROL DCP 0110
C DCP 0111
C READ AND INTERPRET PROCSS CARD DCP 0112
200 CONTINUE DCP 0113
READ (5,5201) NPROC DCP 0114
IF (NPROCS,EQ.0) GO TO 100 DCP 0115
WRITE (6,6201) NPROC DCP 0116
IFDSTT = NPROCS/4 DCP 0117
IPROCS = NPROCS-4*IFDSTT DCP 0118
IF (NPREST.GT.NESFND) NPREST = NESEND DCP 0119
C DCP 0120
C CONVERT PROCESS TIMES TO SECONDS DCP 0121
DO 201 I=1,2 DCP 0122
IPROC(I+3) = 2 DCP 0123
IF (NDATE(2,I).LE.0) GO TO 201 DCP 0124
PDATE(1,I) = 190*NDATE(1,I)+1 DCP 0125
PDATE(2,I) = NDATE(2,I) DCP 0126
PDATE(3,I) = NDATE(3,I) DCP 0127
CALL DATINP (PDATE(1,I),STIMR(I)) DCP 0128
IPROC(I+3) = 1 DCP 0129
201 CONTINUE DCP 0130
C DCP 0131
C POSITION ESTIMATION TAPE DCP 0132
210 IF (NPRFST.LE.0) GO TO 220 DCP 0133
NN = NPREST-NESPOS-1 DCP 0134
IF (NN) 211,215,213 DCP 0135
211 NN = -2*NN DCP 0136
DO 212 I=1,NN DCP 0137
212 BACKSPACE 12 DCP 0138
GO TO 215 DCP 0139
213 DO 214 I=1,NN DCP 0140
READ (12) SKIP DCP 0141
214 READ (12) SKIP DCP 0142
215 CONTINUE DCP 0143
READ (12) CEST DCP 0144
C DCP 0145
C POSITION EDITED DATA TAPE DCP 0146
220 CONTINUE DCP 0147
IF (NPROCS,EQ.1) GO TO 300 DCP 0148
IFRR = 13 DCP 0149

```

DO 221 INDSTA=1,NUMSTA	DCP 0150
221 IF (NPRSTA.EQ.NAMSTA(INDSTA)) GO TO (222,223) ,IFTSTP	DCP 0151
IERR = 12	DCP 0152
GO TO 900	DCP 0153
222 IF (STIMR(2).LT.TREGIN) GO TO 900	DCP 0154
223 IF (STIMR(1).GT.TFND) GO TO 900	DCP 0155
IF (NREWIND.EQ.0) GO TO 224	DCP 0156
REWIND 10	DCP 0157
READ (10) SKIP	DCP 0158
224 IF (NDATF(2,1).LE.0) STIMR(1) = STIMNX(INDSTA)	DCP 0159
225 READ (10) INDDAT	DCP 0160
NM = 1	DCP 0161
NN = 2*NRCO-1	DCP 0162
IF (NEOT) 232,226,227	DCP 0163
226 IF (NRSTA.EQ.NPRSTA) NM = 2	DCP 0164
IF (STIMR(1).LT.ONTIME) GO TO 227	DCP 0165
IF (STIMR(1).EQ.ONTIME) GO TO (227,228) ,NM	DCP 0166
GO TO (230,228) ,NM	DCP 0167
227 NM = 1	DCP 0168
NN = NN+2	DCP 0169
228 DO 229 I=1,NN	DCP 0170
229 BACKSPACE 10	DCP 0171
GO TO (225,231) ,NM	DCP 0172
230 CONTINUE	DCP 0173
READ (10) SKIP	DCP 0174
231 READ (10) INDDAT	DCP 0175
IERR = 14	DCP 0176
IF (NEOT.GT.0) GO TO 900	DCP 0177
IF (STIMR(2).LT.ONTIME) GO TO (900,232) ,IFTSTP	DCP 0178
232 IF (STIMR(1).GT.TLAST) GO TO 230	DCP 0179
IF (NRSTA.NE.NPRSTA) GO TO 230	DCP 0180
IFRR = 0	DCP 0181
READ (10) BUFDAT	DCP 0182
DO 233 ITRSTA=1,NUMSTA	DCP 0183
233 IF (ITRSTA.EQ.NAMSTA(ITRSTA)) GO TO 300	DCP 0184
C	DCP 0185
C**** DATA START	DCP 0186
C	DCP 0187
300 IF (IFDST.EQ.0) GO TO 350	DCP 0188
CALL MLESTT	DCP 0189
IF (IERR.NE.0) GO TO 900	DCP 0190
350 IF (IPROCS.EQ.0) GO TO 200	DCP 0191
GO TO (400,400,500) ,IPROCS	DCP 0192
C	DCP 0193
C**** PROPAGATE/ESTIMATE	DCP 0194
C	DCP 0195
400 CONTINUE	DCP 0196
CALL DIFCOR	DCP 0197
IF (IERR.NE.0) GO TO 900	DCP 0198
GO TO 200	DCP 0199
C	DCP 0200
C**** OUTPUT RESIDUALS	DCP 0201
C	DCP 0202
500 CONTINUE	DCP 0203
CALL RSINDL (NRSPOS)	DCP 0204
IF (IERR.NE.0) GO TO 900	DCP 0205
GO TO 200	DCP 0206
C	DCP 0207
C**** PROGRAM STOPS	DCP 0208
C	DCP 0209
C	DCP 0210
900 ERROR STOP	DCP 0211
CONTINUE	DCP 0212
NN = NERR(IERR)	DCP 0213
NM = NERR(IERR+1)-1	DCP 0214
WRITE (6,6901) (DIAGN(I),I=NN,NM)	DCP 0215
C	DCP 0216
C	DCP 0217
950 NORMAL STOP	DCP 0218
CONTINUE	DCP 0219
WRITE (6,6902) NRSEND,NESEND	
STOP	
END	

Subroutine: DCRØSS

Purpose: To form the double precision cross product
of two vectors.

Calling Sequence: CALL DCRØSS(A,B,C)

Input and Output:

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	A	d(3)			Pre-vector
I	B	d(3)			Post-vector
Ø	C	d(3)			AXB

Common storages used: None

Subroutines required: None

DCRØSS-1

```
$IBFTC MC13M2 XR3,M94,NODD,LIST
SUBROUTINE DCROSS(A,B,C)
DOUBLE PRECISION A,B,C
DIMENSION A(3),B(3),C(3),IN(4)
DATA IN/2,3,1,2/
DO 1 I=1,3
J = IN(I)
K = IN(I+1)
1 C(I) = A(J)*B(K)-A(K)*B(J)
RETURN
END
```

```
DCRS0000
DCRS0001
DCRS0002
DCRS0003
DCRS0004
DCRS0005
DCRS0006
DCRS0007
DCRS0008
```

Subroutine: DDØT

Purpose: Computes the inner product of two 3-vectors in double precision.

Calling Sequence: Z = DDØT (X, Y)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	X	d(3)			Pre-vector.
I	Y	d(3)			Post-vector.
Ø	DDØT	d			Inner product X·Y

Common storages used: None

Subroutines required: None

DDØT-1

```
$IBFTC MC132D XR3,M94,NODD,LIST
  DOUBLE PRECISION FUNCTION DDOT (X,Y)
  DOUBLE PRECISION      X(3),Y(3),SUM
  1 SUM = 0.
  DO 2 I=1,3
  2 SUM = SUM+X(I)*Y(I)
  DDOT = SUM
  RETURN
  END
```

```
DDOT0001
DDOT0002
DDOT0003
DDOT0004
DDOT0005
DDOT0006
DDOT0007
```

Subroutine: DEHA

Purpose: To compute the Greenwich hour angle of the true equinox of date and the Earth's angular velocity.

Calling Sequence: CALL DEHA (T,DA,RA, ϕ M)

Input and Output

I/ ϕ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	T	d	T	sec	Time from 1950 January 0.0 UT
I	DA	d	$\delta\alpha$	rad	Nutation in right ascension
ϕ	RA	d	η	rad	Hour angle of the true equinox of date
ϕ	ϕ M	d	ω	rad/sec	Earth's angular velocity

Common storages used: None

Subroutines required: None

DEHA-1

Method

Universal time is defined as 12^h + Greenwich hour angle of a point on the equator whose longitude is

$$L = 18^h 38^m 45^s.836 + 8640184^s.542 \hat{T}_u + 0^s.0929 \hat{T}_u^2$$

measured from the mean equinox of date, where \hat{T}_u is measured in Julian centuries from 1900 Jan 0.5 UT. Setting T_u = Julian centuries from 1950 Jan 0.0 UT,

$$L-12^h = 1.746\ 647\ 719\ 162\ 569 + 0.0172\ 027\ 914\ 509\ 729\ 83 [d_u] \\ + 0.506\ 408\ 971\ 100224\ 10 \times 10^{-14} [d_u]^2 + \omega t \text{ radians}$$

$$[d_u] = \text{integer part of } [36525 T_u]$$

= whole days since 1950 Jan 0.0 UT

t = seconds from 0^h of date

ω = Earth's angular velocity

$$= \left\{ (L-12^h) \frac{[d_u] + 1}{[d_u]} + 2\pi \right\} / 86400 \text{ radians/second}$$

The G.H.A. of the true vernal equinox, φ , is

$$\varphi = L-12^h + \delta\alpha$$

where $\delta\alpha$ is the nutation in right ascension.

DEHA-2

```

$IBFTC MC132Y XR3,M94,NODD,LIST
SUBROUTINE DEHA (T,DA,RA,OM)
C   HOUR ANGLE OF THE TRUE EQUINOX
    DOUBLE PRECISION T,DA,RA,OM
    1      ,C(7),DU,TS,DMOD
DATA   C/1.746647719162569 ,1.72027914509729 D-02
1      ,5.064089711002241D-15,7.292115854896487D-05
2      ,1.172242988657926D-19,6.283185307179586
3      ,8.64          D+04/
C
1 DU = AINT(T/C(7))
  TS = T-C(7)*DU
  OM = C(4)+C(5)*DU
  TS = DA+OM*TS+C(1)+DU*(C(2)+DU*C(3))
  RA = DMOD(TS,C(6))
999 RETURN
    END

```

```

DEHA0001
DEHA0002
DEHA0003
DEHA0004
DEHA0005
DEHA0006
DEHA0007
DEHA0008
DEHA0009
DEHA0010
DEHA0011
DEHA0012
DEHA0013
DEHA0014
DEHA0015

```

Subroutine: DELEV

Purpose: To sort the critical event array in ascending order and determine the measurement interval DEL.

Calling Sequence: CALL DELEV (EVNT,KEV,DEL,KSTØP,NSTA)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
IØ	EVNT	(6)		seconds	Critical event times. All except the first must be in ascending order on entry.
IØ	KEV	(6)			Critical event keys; KEV(I) is the key for EVNT(I).
Ø	DEL			seconds	Measurement interval.
Ø	KSTØP				Key indicating type of first of ordered output critical events. KSTØP = KEV(1)
I	NSTA				Number of station having earliest next on time.

Common storages used: /INPCØM/, /WCØM/

Subroutines required: None

DELEV-1

\$IBFTC MC13DL	NOREF,M94,NODD,XR3	
CMC13DL	DELEV	EVDEL WITH DIFFERENT LOGIC FOR FINDING DEL
	SUBROUTINE	DELEV(EVNT,KEV,DEL,KSTOP,NSTA)
	COMMON	/INPCOM/C(700)/WCOM/IW(550),CW(1450)
	DIMENSION	EVNT(6), KEV(6), S(23,12), ISC(12)
	EQUIVALENCE	(C(200),S), (C(556),DELBE), (C(573),DELRAD)
	1,	(C(580),DELOPT), (IW(8),ISC)
	IF(EVNT(1).LE.EVNT(2))	GO TO 4
	ESAV=EVNT(1)	
	KSAV=KEV(1)	
	DO 1 J=1,4	
	I=J+2	
	IF(ESAV-EVNT(I))	2,2,1
1	CONTINUE	
	J=J+1	
2	CONTINUE	
	DO 3 I=1,J	
	EVNT(I)=EVNT(I+1)	
3	KEV(I)=KEV(I+1)	
	J=J+1	
	EVNT(J)=ESAV	
	KEV(J)=KSAV	
4	CONTINUE	
	KSTOP=KEV(1)	
	GO TO (20,30,40,50,100,100),KSTOP	
20	N=ISC(NSTA)	
	DEL=S(2,N)	
	GO TO 100	
30	DEL=DELBE	
	GO TO 100	
40	DEL=DELOPT	
	GO TO 100	
50	DEL=DELRAD	
100	RETURN	
	END	

DELV0001
 DELV0002
 DELV0003
 DELV0004
 DELV0005
 DELV0006
 DELV0007
 DELV0008
 DELV0009
 DELV0010
 DELV0011
 DELV0012
 DELV0013
 DELV0014
 DELV0015
 DELV0016
 DELV0017
 DELV0018
 DELV0019
 DELV0020
 DELV0021
 DELV0022
 DELV0023
 DELV0024
 DELV0025
 DELV0026
 DELV0027
 DELV0028
 DELV0029
 DELV0030
 DELV0031
 DELV0032
 DELV0033
 DELV0034

Subroutine: DEQD

Purpose: Integrates second-order differential equations in double precision.

Calling Sequence: CALL DEQD(FSUB,ØSUB)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
	FSUB				Name of subroutine which evaluates derivatives
	ØSUB				Name of subroutine which controls output.

Common storages used: //(40),/DQDCØM/

Subroutines required: FSUB,ØSUB

DEQD-1

Restrictions

An "EXTERNAL FSUB,ØSUB" statement must be included in the calling program. The subroutines FSUB,ØSUB must have zero length call lists. For use with ØVERLAY, FSUB,ØSUB and the labelled common DQDCØM must be included in the link containing DEQD or in a higher link.

The functions of these subroutines are

FSUB Computes the second derivatives of the dependent variables, and any stopping functions.

ØSUB Provides for output at completion of integration of each interval (regular points) and at certain points obtained by interpolation.

The Kutta mode may not be used if any interpolation is required (i.e., even-interval output or stopping functions).

Usage

DEQD is designed to integrate the second-order differential equations

$$\frac{d}{dx} (RDEQ(i)) = VDEQ(i)$$

$$\frac{d}{dx} (VDEQ(i)) = ADEQ(i)$$

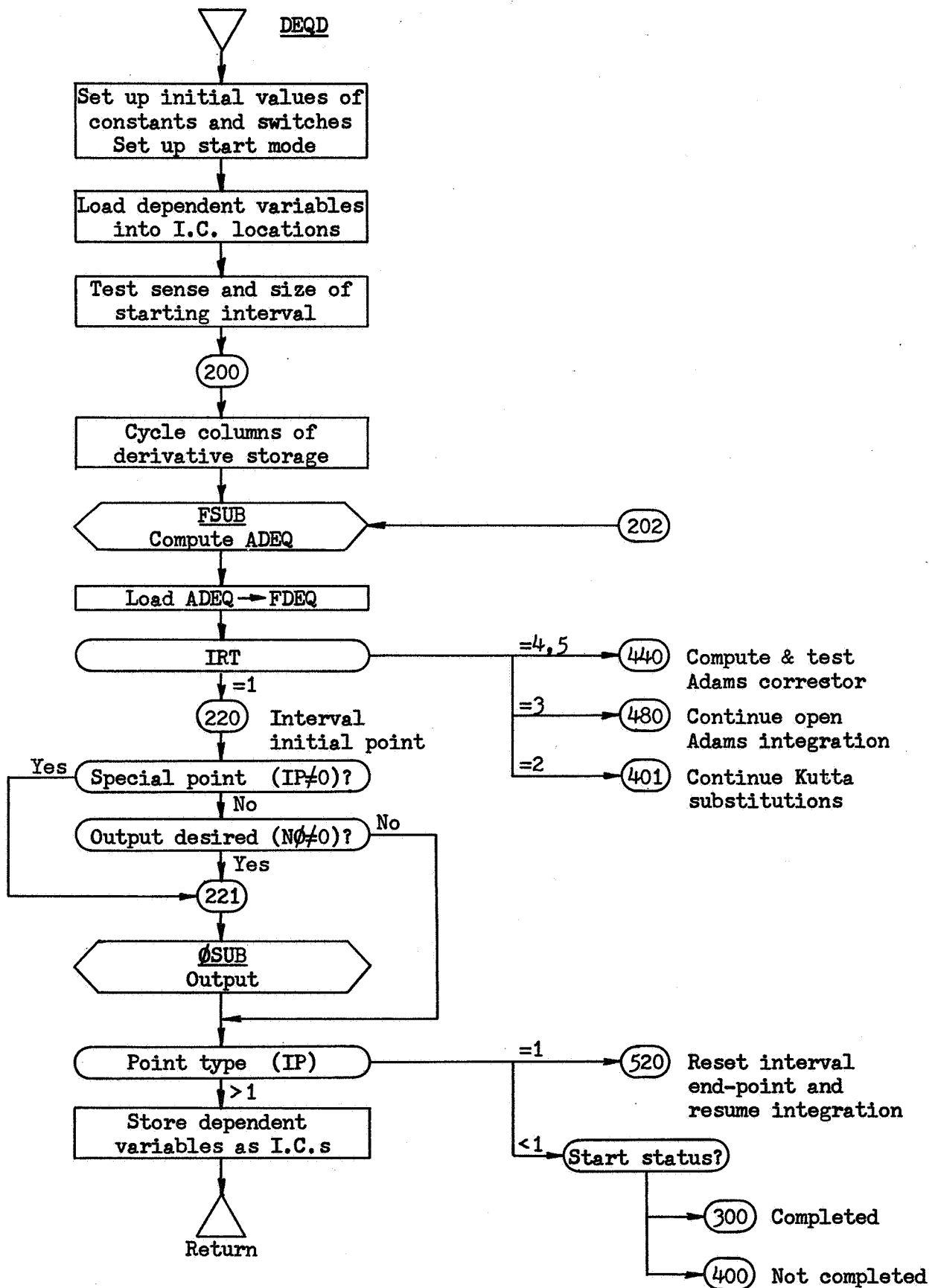
from initial conditions pre-loaded into RDEQ,VDEQ, and derivatives loaded into ADEQ by the subroutine "FSUB". NE upper-loaded equations are integrated from the pre-loaded $x = XDEQ(1)$ to the specified final point $XDEQ(2)$.

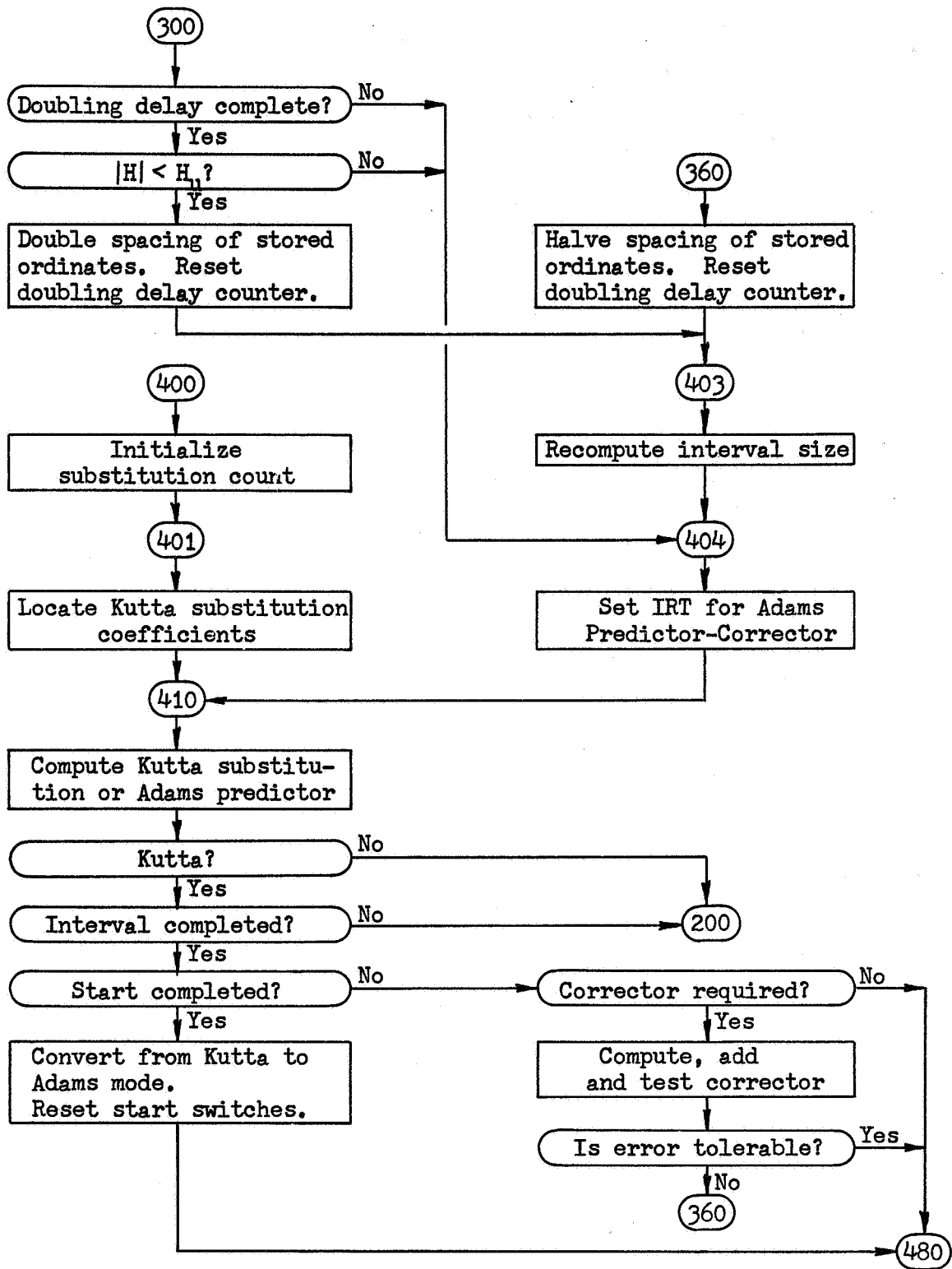
If $NS \neq 0$, NS functions are loaded into ADEQ by "FSUB", starting at ADEQ(NE+1). The first zero of these functions, if any, causes a return from DEQD with XDEQ(1) at the zero, and IP = 3. The function causing the return is ADEQ(IT).

Interpolated values of RDEQ,VDEQ may be obtained at even intervals for output purposes during Adams integration. The first point interpolated is the pre-loaded XDEQ(3) and others follow at the interval HE.

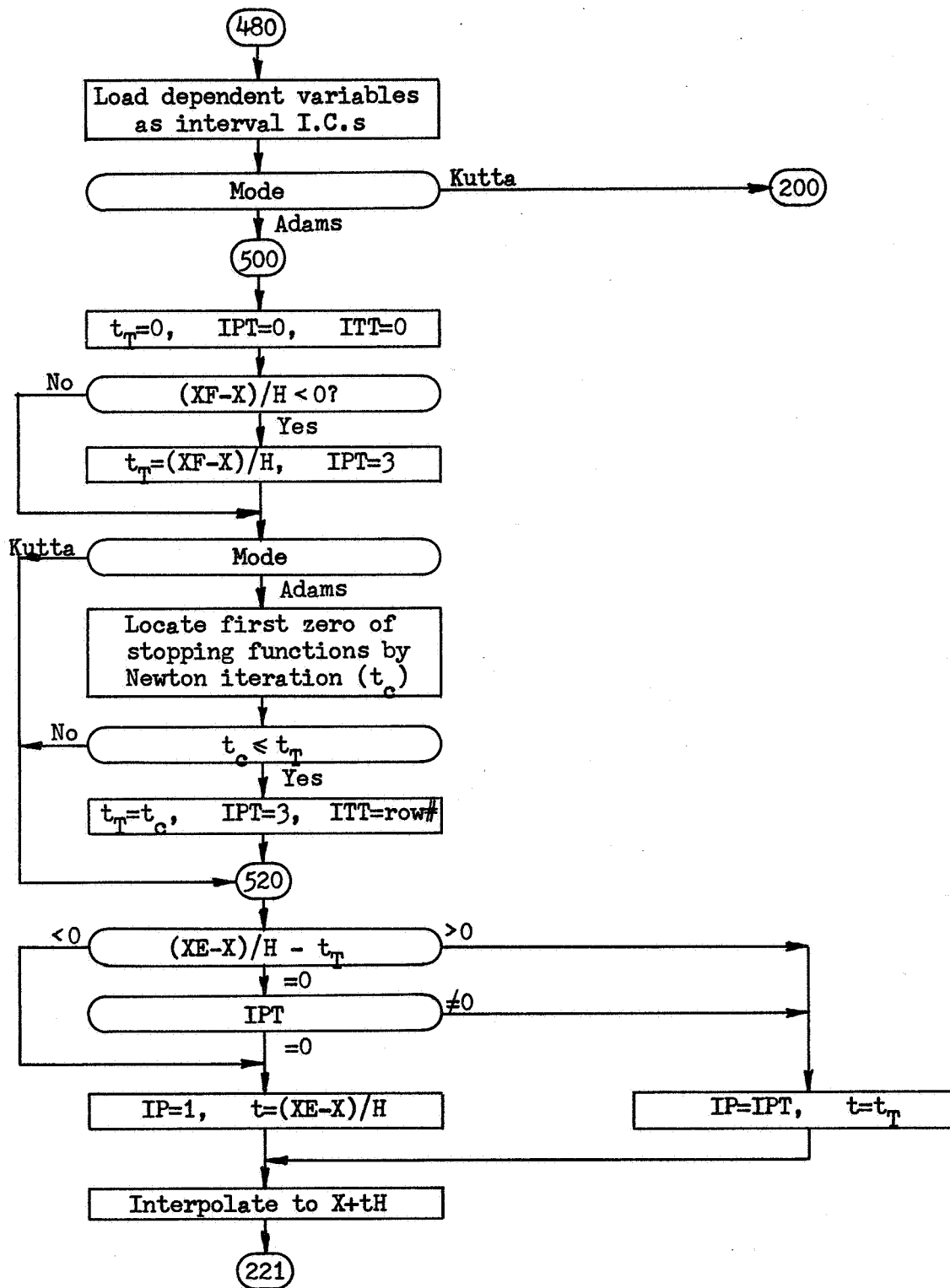
The integration, interpolation, and accuracy formulas are described in Appendix D of Reference 1.

DEQD-3





DEQD-5



DEQD-6

```

SIBFTC MC133D XR3,M94,NODD,LIST
SUBROUTINE DEQD (FSUB,OSUB)
C   DEQD - DOUBLE PRECISION INTEGRATION
C   OF SECOND ORDER EQUATIONS
C
COMMON      /DQDCOM/NDEQ(10),CDEQ(10),XDEQ (4)
1           ,ADEQ(44),RDEQ(44),VDEQ(44),FDEQ(44,10)
DOUBLE PRECISION  ADEQ,FDEQ,RDEQ,VDEQ,XDEQ
1           ,X,XF,XE,XO
EQUIVALENCE    (NDEQ( 1),NI),(CDEQ( 1),H ),(XDEQ( 1),X )
1           ,(NDEQ( 2),NO),(CDEQ( 2),HP),(XDEQ( 2),XF)
2           ,(NDEQ( 3),NS),(CDEQ( 3),HO),(XDEQ( 3),XE)
3           ,(NDEQ( 4),NA),(CDEQ( 4),HU),(XDEQ( 4),XO)
4           ,(NDEQ( 5),NC),(CDEQ( 5),HL)
5           ,(NDEQ( 6),NE),(CDEQ( 6),HE)
6           ,(NDEQ( 7),NT),(CDEQ( 7),EU)
7           ,(NDEQ( 8),IP),(CDEQ( 8),EL)
8           ,(NDEQ( 9),IT)
C
COMMON      STP(40)
DOUBLE PRECISION  TMP(20)
EQUIVALENCE    (STP(1),TMP)
C
DOUBLE PRECISION  ALP(4),CQD(12),CQX(4),DCN(2),DX,FLT(7)
DIMENSION      CAE (7),CDB(7,2),CQK (20),SFL (14)
1           ,CAI(6,6),CHV(6,3),IN (8),SLP (8)
2           ,CCV(8,6),CQA (12)
EQUIVALENCE    (ALP(1),SLP),(FLT(1),SFL)
C
DATA CAE/ 1., -6., 15.,-20., 15., -6., 1./
DATA CAI/5*0.,60., -12., 75., -200., 300., -300., 137.
1           , -50., 305., -780., 1070., -770., 225.
2           , -105., 615.,-1470., 1770.,-1065., 255.
3           , -120., 660.,-1440., 1560., -840., 180.
4           , -60., 300., -600., 600., -300., 60./
DATA CCV/ 3864.,-3200.,-6496.,-1539.,12327.,-4300., -224., 324.
1           , 1176., -400.,-3584., -972., 7812.,-3600., 0., 324.
2           , 126., 300.,-1344., -405., 3717.,-1800., 0., 162.
3           , 0., 0., 0., 0., 756., 0., 0., 0.
4           , 84., -200., 224., 81., -357., 700., 224., 0.
5           , -336., 800., -896., -324., 1092., -800., 896., 324./
DATA CDB/ -224., -112., -48., 28., 210., 140., 7.
1           , -240.,-1008., -560., 210., 2100., 1575., 84./
DATA CHV/-0.01171875, 0.08203125,-0.2734375 , 0.8203125
1           , 0.41015625,-0.02734375, 0.01171875,-0.09865625
2           , 0.5859375 , 0.5859375 , -0.09865625, 0.01171875
3           , 0.02737375,-0.17578125, 0.4921875 , -0.8203125
4           , 1.23046875, 0.24609375/
DATA CQA/ -863., 5260.,-13474., 18752.,-15487., 10852.
1           , -3325., 20139.,-51086., 69874.,-55461., 29939./
DATA CQK/ 2268., 0., 2268., 1680., -196., 784., 420.
1           , 1400., 448., 0., 4536., -3969., 8505., 13944.
2           , -15680., 6272., 420., 2000., 1792., 324./
DATA CQX/ 1.0D+0, 1.0D+0, .75D+0, .30D+0/
DATA DCN/ 3.168648818567611, 3.966129870129870/
DATA FLT/ 1.D0, 2.D0, 3.D0, 4.D0, 5.D0, 6.D0, 7.D0/
C
C**** PROLOGUE          SET FIXED DATA AND SWITCHES
C
100 DO 101 I=1,8
101 IN(I) = I
    HP = 0.
    EL = EU/128.
    ASSIGN 234 TO LLA
    ASSIGN 480 TO LLB
    ASSIGN 480 TO LLC
    ASSIGN 200 TO LLD
    ASSIGN 521 TO LLE
    ASSIGN 200 TO LLF
    IF (NI.EQ.0) GO TO 103
    IF (NS.NE.0) ASSIGN 502 TO LLE
    IF (NA.NE.0) ASSIGN 450 TO LLC
    ASSIGN 202 TO LLF
    IF (NC.NE.2) ASSIGN 220 TO LLF
103 IRT = 1

```

```

DEQD0001
DEQD0002
DEQD0003
DEQD0004
DEQD0005
DEQD0006
DEQD0007
DEQD0008
DEQD0009
DEQD0010
DEQD0011
DEQD0012
DEQD0013
DEQD0014
DEQD0015
DEQD0016
DEQD0017
DEQD0018
DEQD0019
DEQD0020
DEQD0021
DEQD0022
DEQD0023
DEQD0024
DEQD0025
DEQD0026
DEQD0027
DEQD0028
DEQD0029
DEQD0030
DEQD0031
DEQD0032
DEQD0033
DEQD0034
DEQD0035
DEQD0036
DEQD0037
DEQD0038
DEQD0039
DEQD0040
DEQD0041
DEQD0042
DEQD0043
DEQD0044
DEQD0045
DEQD0046
DEQD0047
DEQD0048
DEQD0049
DEQD0050
DEQD0051
DEQD0052
DEQD0053
DEQD0054
DEQD0055
DEQD0056
DEQD0057
DEQD0058
DEQD0059
DEQD0060
DEQD0061
DEQD0062
DEQD0063
DEQD0064
DEQD0065
DEQD0066
DEQD0067
DEQD0068
DEQD0069
DEQD0070
DEQD0071
DEQD0072
DEQD0073
DEQD0074

```

```

      IRS = 0
      IPT = 0
      IP = 2
      H = 0.
      ALP(4) = 0.D0
      GO TO 480
140  H = H0
      TMP(1) = XF-X
      IF (STP(1).LT.0.) H=-H
      IF (STP(1)/H.GT.2.) GO TO 141
      H = TMP(1)/2.D0
      IPT = 3
141  IP = 3
      IF (ABS(H).EQ.0.) GO TO 221
      IP = 0
      IT = 0
      NI = NE+1
      NT = NE+NS
C
C**** CONTROL          SELECT COMPUTATIONAL PATH
C
C      COMPUTE AND LOAD DERIVATIVES
200  I = IN(1)
      DO 201 J=1,7
201  IN(J) = IN(J+1)
      IN(8) = I
202  CALL FSUB
      I = IN(8)
      DO 203 J=1,NT
203  FDEQ(J,I) = ADEQ(J)
      GO TO (220,401,480,440) ,IRT
C
C      OUTPUT CONTROL
220  IF (IP.NE.0) GO TO 221
      IF (NO.EQ.0) GO TO 233
221  CALL OSUB
C
C      IDENTIFY POINT TYPE AND SELECT PATH
230  IF (IP) 231,232,232
231  IPT = 0
      NS = 0
      NT = NE
      XF = XF+FLT(7)*DX
      GO TO 300
232  IF (IP-1) 233,520,480
233  GO TO LLA, (234,400,300)
234  IF (HP.EQ.0.) GO TO 400
      ASSIGN 400 TO LLA
      IF (NI.EQ.0) GO TO 400
      ASSIGN 300 TO LLA
      ASSIGN 420 TO LLB
      ASSIGN 500 TO LLD
      GO TO 400
C
C**** NEXT          INTERVAL SIZE CONTROL
C
C      SELECT NEXT INTERVAL SIZE
300  IF (IDL.GT.0) GO TO 301
      IF (ABS(H).GE.HU) GO TO 404
      GO TO 340
301  IDL = IDL-1
      GO TO 404
C
C      DOUBLE SPACING OF ADAMS ORDINATES
340  H = H*SFL(3)
      I = IN(7)
      J = IN(5)
      IN(7) = IN(6)
      IN(6) = IN(4)
      IN(5) = IN(2)
      IN(4) = I
      IN(2) = J
      DO 342 I=1,NT
      TMP(1) = 0.D0
      TMP(2) = 0.D0

```

```

DEQD0075
DEQD0076
DEQD0077
DEQD0078
DEQD0079
DEQD0080
DEQD0081
DEQD0082
DEQD0083
DEQD0084
DEQD0085
DEQD0086
DEQD0087
DEQD0088
DEQD0089
DEQD0090
DEQD0091
DEQD0092
DEQD0093
DEQD0094
DEQD0095
DEQD0096
DEQD0097
DEQD0098
DEQD0099
DEQD0100
DEQD0101
DEQD0102
DEQD0103
DEQD0104
DEQD0105
DEQD0106
DEQD0107
DEQD0108
DEQD0109
DEQD0110
DEQD0111
DEQD0112
DEQD0113
DEQD0114
DEQD0115
DEQD0116
DEQD0117
DEQD0118
DEQD0119
DEQD0120
DEQD0121
DEQD0122
DEQD0123
DEQD0124
DEQD0125
DEQD0126
DEQD0127
DEQD0128
DEQD0129
DEQD0130
DEQD0131
DEQD0132
DEQD0133
DEQD0134
DEQD0135
DEQD0136
DEQD0137
DEQD0138
DEQD0139
DEQD0140
DEQD0141
DEQD0142
DEQD0143
DEQD0144
DEQD0145
DEQD0146
DEQD0147
DEQD0148
DEQD0149

```

```

DO 341 J=1,7
L = IN(J+1)
TMP(1) = TMP(1)+CDB(J,1)*FDEQ(I,L)
341 TMP(2) = TMP(2)+CDB(J,2)*FDEQ(I,L)
L = IN(4)
FDEQ(I,L) = TMP(1)
L = IN(3)
342 FDEQ(I,L) = TMP(2)
GO TO 403
C
C HALVE SPACING OF ADAMS ORDINATES
360 H = H/SFL(3)
DO 363 I=1,NT
DO 361 J=1,6
361 STP(J) = 0.
DO 362 J=1,6
L = IN(J+1)
TMP(4) = FDEQ(I,L)
DO 362 K=1,3
362 TMP(K) = TMP(K)+TMP(4)*CHV(J,K)
DO 363 J=1,3
L = IN(J+1)
363 FDEQ(I,L) = TMP(J)
I = IN(8)
IN(8) = IN(7)
IN(7) = IN(4)
IN(4) = IN(5)
IN(5) = IN(2)
IN(2) = I
GO TO 403
C
C**** QUADRATURE INTEGRATE TO X0+H
C
C LOCATE KUTTA SUBSTITUTION COEFFICIENTS
400 IRT = 2
ISB = 9
ICN = 1
401 ISB = ISB-1
DX = H*CQX(ISB-4)
DO 402 I=ISB,8
CQD(I-2) = CQK(ICN)/4536.D0
CQD(I+4) = CQK(ICN+10)/4536.D0
402 ICN = ICN+1
GO TO 410
403 IDL = 2
DX = H
ALP(2) = 60.D0/DX
ALP(3) = ALP(2)/DX
404 IRT = NRT
C
C INTEGRATE/PREDICT
410 X = X0+DX
DO 413 I=1,NE
TMP(1) = 0.D0
TMP(2) = 0.D0
DO 411 J=ISB,8
L = IN(J)
TMP(1) = TMP(1)+CQD(J-2)*FDEQ(I,L)
411 TMP(2) = TMP(2)+CQD(J+4)*FDEQ(I,L)
DO 412 J=1,2
412 TMP(J) = FDEQ(I,10)+DX*TMP(J)
RDEQ(I) = FDEQ(I,9)+DX*TMP(1)
413 VDEQ(I) = TMP(2)
IF (IRT.NE.2) GO TO 200
IF (ISB.GT.5) GO TO 200
GO TO LLB, (480,420,440,432)
C
C**** CONVERT CHANGE FROM START TO ADAMS MODE
C
C COMPUTE ADAMS ORDINATES
420 H = H/SFL(3)
DO 422 I=1,NT
DO 421 J=1,6
TMP(J) = 0.D0
DO 421 L=1,8

```

```

DEQD0150
DEQD0151
DEQD0152
DEQD0153
DEQD0154
DEQD0155
DEQD0156
DEQD0157
DEQD0158
DEQD0159
DEQD0160
DEQD0161
DEQD0162
DEQD0163
DEQD0164
DEQD0165
DEQD0166
DEQD0167
DEQD0168
DEQD0169
DEQD0170
DEQD0171
DEQD0172
DEQD0173
DEQD0174
DEQD0175
DEQD0176
DEQD0177
DEQD0178
DEQD0179
DEQD0180
DEQD0181
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DEQD0192
DEQD0193
DEQD0194
DEQD0195
DEQD0196
DEQD0197
DEQD0198
DEQD0199
DEQD0200
DEQD0201
DEQD0202
DEQD0203
DEQD0204
DEQD0205
DEQD0206
DEQD0207
DEQD0208
DEQD0209
DEQD0210
DEQD0211
DEQD0212
DEQD0213
DEQD0214
DEQD0215
DEQD0216
DEQD0217
DEQD0218
DEQD0219
DEQD0220
DEQD0221
DEQD0222
DEQD0223
DEQD0224

```

421	TMP(J) = TMP(J)+CCV(L,J)*FDEQ(I,L)	DEQD0225
	DO 422 J=3,8	DEQD0226
422	FDEQ(I,J) = TMP(J-2)/756.DO	DEQD0227
C		DEQD0228
C	SET ADAMS INTEGRATION PARAMETERS	DEQD0229
430	ISB = 3	DEQD0230
	NRT = 3	DEQD0231
	ASSIGN 480 TO LLB	DEQD0232
	DO 431 I=1,12	DEQD0233
431	CQD(I) = CGA(I)/10080.DO	DEQD0234
	IF (NA+NC.EQ.0) GO TO 480	DEQD0235
	NRT = 4	DEQD0236
	IDL = 4	DEQD0237
	IP = IPT	DEQD0238
	ASSIGN 432 TO LLB	DEQD0239
	IF (IP.NE.0) GO TO 221	DEQD0240
	GO TO 480	DEQD0241
432	IF (IDL.GT.1) GO TO 480	DEQD0242
	ASSIGN 440 TO LLB	DEQD0243
C		DEQD0244
C****	CORRECT TEST ACCURACY AND ADD ADAMS CORRECTOR	DEQD0245
C		DEQD0246
C	ADD CORRECTOR AND COMPUTE MAXIMUM	DEQD0247
440	STP(1) = 0.	DEQD0248
	DO 442 I=1,NE	DEQD0249
	STP(2) = ABS(VDEQ(I))	DEQD0250
	IF (SFL(1).GT.STP(2)) STP(2) = SFL(1)	DEQD0251
	TMP(3) = 0.DO	DEQD0252
	DO 441 J=1,7	DEQD0253
	L = IN(J+1)	DEQD0254
441	TMP(3) = TMP(3)+CAE(J)*FDEQ(I,L)	DEQD0255
	TMP(3) = DX*TMP(3)/DCN(1)	DEQD0256
	TMP(2) = DX*TMP(3)/DCN(2)	DEQD0257
	STP(2) = ABS(STP(5))/STP(2)	DEQD0258
	IF (STP(2).GT.STP(1).AND.I.LE.3) STP(1) = STP(2)	DEQD0259
	IF (NC.EQ.0) GO TO 442	DEQD0260
	RDEQ(I) = RDEQ(I)+TMP(2)	DEQD0261
	VDEQ(I) = VDEQ(I)+TMP(3)	DEQD0262
442	CONTINUE	DEQD0263
	GO TO LLC, (480,450)	DEQD0264
C		DEQD0265
C	TEST ACCURACY	DEQD0266
450	IF (STP(1).LT.EL) GO TO 480	DEQD0267
	IDL = IDL+1	DEQD0268
	IF (STP(1).LT.EU) GO TO 480	DEQD0269
	IF (ABS(H).LE.HL) GO TO 480	DEQD0270
	GO TO 360	DEQD0271
C		DEQD0272
C****	ADVANCE STORE ICS FOR NEXT INTERVAL	DEQD0273
C		DEQD0274
480	IRT = 1	DEQD0275
	HP = H	DEQD0276
	DO 481 I=1,NE	DEQD0277
	FDEQ(I, 9) = RDEQ(I)	DEQD0278
481	FDEQ(I,10) = VDEQ(I)	DEQD0279
	X0 = X	DEQD0280
	IF (IP-2) 482,140,999	DEQD0281
482	GO TO LLD, (500,200)	DEQD0282
C		DEQD0283
C****	TEST LOCATE AND ORDER SPECIAL POINTS	DEQD0284
C		DEQD0285
C	FIND STOPPING POINTS	DEQD0286
500	ALP(4) = 0.DO	DEQD0287
	IPT = 0	DEQD0288
	ITT = 0	DEQD0289
	ALP(1) = (XF-X)/DX	DEQD0290
	IF (SLP(1).GT.0.) GO TO 501	DEQD0291
	ALP(4) = ALP(1)	DEQD0292
	IPT = 3	DEQD0293
501	GO TO LLE, (502,521)	DEQD0294
502	K = IN(8)	DEQD0295
	L = IN(7)	DEQD0296
	DO 512 I=N1,NT	DEQD0297
	TMP(2) = FDEQ(I,L)-FDEQ(I,K)	DEQD0298
	M = 3	DEQD0299

<pre> GO TO 504 503 K = IN(6) M = 2 TMP(2) = (TMP(2)+FDEQ(I,K)-FDEQ(I,L))/FLT(2) 504 IF (STP(3).EQ.0.) GO TO 506 TMP(1) = FDEQ(I,L)/TMP(2)-FLT(1) 505 IF (STP(1).GT.SLP(7)) GO TO 506 IF (STP(1)+SFL(3).LE.0.) GO TO 506 GO TO (512,507,507) ,M 506 GO TO (512,512,503) ,M 507 M = 1 DO 508 J=1,12 508 STP(J+6) = 0. DO 509 J=1,6 L = IN(J+2) TMP(10) = FDEQ(I,L) DO 509 K=1,6 509 TMP(K+3) = TMP(K+3)+TMP(10)*CAI(J,K) DO 510 J=1,5 K = 5-J 510 TMP(9) = TMP(K+4)+TMP(9)*TMP(1)/FLT(K+1) TMP(3) = TMP(9)/TMP(2)/60.D0 TMP(1) = TMP(1)+TMP(3) IF (ABS(STP(5)).GT..1E-11) GO TO 505 IF (TMP(1)-ALP(4)) 511,511,512 511 IPT = 3 ITT = I ALP(4) = TMP(1) 512 CONTINUE GO TO 521 C C FIND EVEN-INTERVAL POINT 520 IRS = 1 521 ALP(1) = (XE-X)/DX IF (ALP(1)-ALP(4)) 523,522,524 522 IF (IPT.NE.0) GO TO 524 523 IP = 1 XE = XE+HE GO TO 525 524 IP = IPT IT = ITT ALP(1) = ALP(4) 525 IF (SLP(1).NE.0.) GO TO 610 IF (IPT.NE.0) GO TO 221 IF (IRS) 600,600,540 C C**** RESET RESTORE INTERVAL END-POINT C 540 IRS = 0 X = X0 L = IN(8) DO 541 I=1,NE RDEQ(I) = FDEQ(I, 9) 541 VDEQ(I) = FDEQ(I,10) DO 542 I=1,NT 542 ADEQ(I) = FDEQ(I,L) C C**** INTERPOLATE COMPUTE INTEGRALS AT X0+T*H C 600 IF (SLP(1).EQ.0.) GO TO LLF, (200,202,220) C C COMPUTE INTERPOLATION COEFFICIENTS 610 TMP(1) = ALP(1) DO 611 I=1,6 611 TMP(I+1) = TMP(1)*TMP(I)/FLT(I+1) DO 612 I=17,40 612 STP(I) = 0. DO 614 I=1,6 DO 613 J=1,6 TMP(8) = CAI(I,J) TMP(I+ 8) = TMP(I+ 8)+TMP(8)*TMP(J) 613 TMP(I+14) = TMP(I+14)+TMP(8)*TMP(J+1) TMP(I+ 8) = TMP(I+ 8)/ALP(2) 614 TMP(I+14) = TMP(I+14)/ALP(3) C </pre>	<pre> DEQD0300 DEQD0301 DEQD0302 DEQD0303 DEQD0304 DEQD0305 DEQD0306 DEQD0307 DEQD0308 DEQD0309 DEQD0310 DEQD0311 DEQD0312 DEQD0313 DEQD0314 DEQD0315 DEQD0316 DEQD0317 DEQD0318 DEQD0319 DEQD0320 DEQD0321 DEQD0322 DEQD0323 DEQD0324 DEQD0325 DEQD0326 DEQD0327 DEQD0328 DEQD0329 DEQD0330 DEQD0331 DEQD0332 DEQD0333 DEQD0334 DEQD0335 DEQD0336 DEQD0337 DEQD0338 DEQD0339 DEQD0340 DEQD0341 DEQD0342 DEQD0343 DEQD0344 DEQD0345 DEQD0346 D5QD0347 DEQD0348 DEQD0349 DEQD0350 DEQD0351 DEQD0352 DEQD0353 DEQD0354 DEQD0355 DEQD0356 DEQD0357 DEQD0358 DEQD0359 DEQD0360 DEQD0361 DEQD0362 DEQD0363 DEQD0364 DEQD0365 DEQD0366 DEQD0367 DEQD0368 DEQD0369 DEQD0370 DEQD0371 DEQD0372 DEQD0373 DEQD0374 </pre>
--	---

```

C INTERPOLATE TO X0+T*H
620 TMP(6) = DX*TMP(1)
X = X0+TMP(6)
DO 621 I=1,NE
VDEQ(I) = FDEQ(I,10)
RDEQ(I) = FDEQ(I, 9)+DX*FDEQ(I,10)
DO 621 J=1,6
L = IN(J+2)
VDEQ(I) = VDEQ(I)+FDEQ(I,L)*TMP(J+ 8)
621 RDEQ(I) = RDEQ(I)+FDEQ(I,L)*TMP(J+14)
GO TO 221
C
C**** RETURN STOP POINT REACHED
C
999 RETURN
END

```

```

DEQD0375
DEQD0376
DEQD0377
DEQD0378
DEQD0379
DEQD0380
DEQD0381
DEQD0382
DEQD0383
DEQD0384
DEQD0385
DEQD0386
DEQD0387
DEQD0388
DEQD0389

```

Subroutine: DEQTR

Purpose: To compute parameters describing the precession of the mean equator of date.

Calling Sequence: CALL DEQTR (T,A)

Input and Output

I/ ϕ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Description
I	T	d		sec	Time from 1950 Jan 0.0 (ET)
ϕ	A(1)	d	$\cos\alpha$		Cosine of precession angle
ϕ	A(2)	d	$\sin\alpha$		Sine of precession angle
ϕ	A(3)	d	$\bar{\epsilon}$		Mean obliquity of the ecliptic
ϕ	A(4-6)	d(3)	w		Vector along the axis of precession.

Common storages used: None

Subroutines required: None

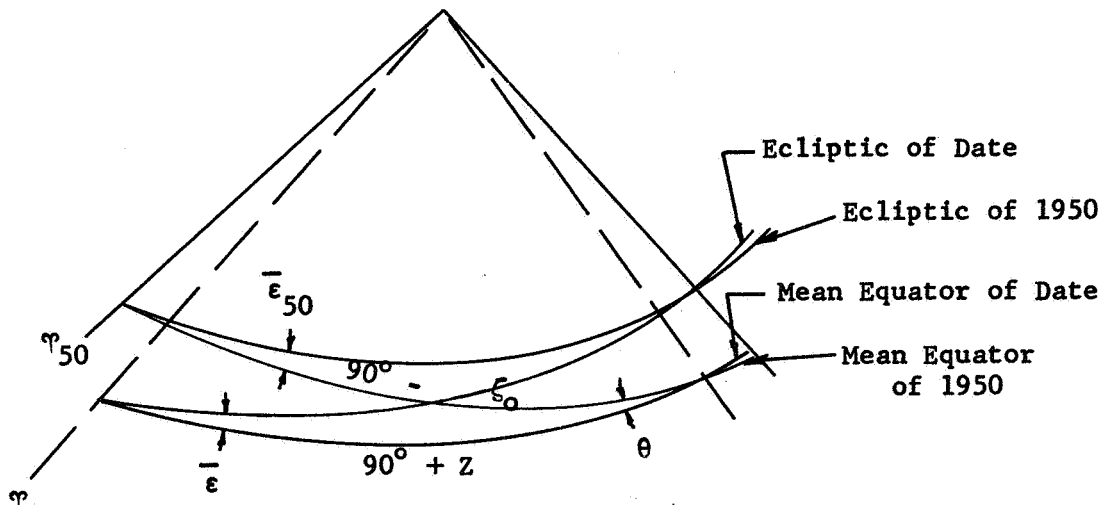
DEQTR-1

Method

The precession of the Earth's mean equator of date is described by the three small angles ζ_0 , Z , θ , where

$$\zeta_0 + Z = \text{precession in right ascension}$$

$$\theta = \text{precession in declination}$$



Precession of the Mean Equator

Numerical values for ζ_0 , Z , θ are given by reference 3, p. 30, in tropical centuries (of 36524.21988 ephemeris days) from 1900.0 (1900 Jan 0.814 ET). The base coordinate system is equator and equinox of 1950.0 (1949 Dec 30.924 ET), and the corresponding expressions in Julian centuries from T_{E1} are

DEQTR-2

T_E = Julian centuries from T_{E1}

$$T = T_E + 0.20824 \times 10^{-5}$$

$$\zeta_o = 2304''.997T + 0''.302T^2 + 0''.018T^3$$

$$Z = 2304''.997T + 1''.093T^2 + 0''.018T^3$$

$$\theta = 2004''.298T - 0''.426T^2 - 0''.042T^3$$

The mean obliquity of the ecliptic, $\bar{\epsilon}$, from reference 3, p.38, is

$$\bar{\epsilon} = 23''.4457878 - 0''.01301376T_E - 0''.8855 \times 10^{-6}T_E^2 + 0''.503 \times 10^{-1}T_E^3$$

The transformation from mean EE of 1950.0 (C-frame) to mean EE of date (D_o -frame) is

$${}^T_{C2D_o} = T_W(\alpha) = T_3(-90^\circ - 2) T_1(\theta) T_3(90^\circ - \zeta_o)$$

Since retention of orthogonality in double precision calculation of ${}^T_{C2D_o}$ from the expansion of the right-hand side requires large polynomials in T (through terms in T^3), the eigenvector, W, and rotation angle, α , together with the mean obliquity are computed by DEQTR, using

$$\sin \alpha = .0243708730T + .65233 \times 10^{-5}T^2 - .493T^3$$

$$\cos \alpha = \sqrt{1 - \sin^2 \alpha}$$

$$W_1 = -.83384 \times 10^{-5}T^2 + .41 \times 10^{-8} T^3$$

$$W_2 = .4347723663 - .2119164 \times 10^{-3}T + .12169 \times 10^{-5}T^2 + .32 \times 10^{-8}T^3$$

$$W_3 = -1.$$

DEQTR-3

SIBFTC MC132Z XR3,M94,NODD,LIST
SUBROUTINE DEQTR (T,A)

C PRECESSION OF THE MEAN EQUATOR
DOUBLE PRECISION T,A(6),C(4,4),DSQRT
DATA C/-.493 D-7, .65233D-5, .24370873 D-1,0.D0
1 , .8779D-8,-.15455D-7,-.227133008D-3, .409206191890
2 , .41 D-8,-.83384D-5,0.D0 ,0.D0
3 , .32 D-8, .12169D-5,-.2119164 D-3, .4347723663/

C
1 TC = T/3.15576D+9+0.20824D-05
DO 2 I=1,4
2 A(I+1) = C(1,I)
DO 3 I=2,4
DO 3 J=1,4
3 A(J+1) = C(I,J)+TC*A(J+1)
A(1) = DSQRT(1.D0-A(2)*A(2))
A(6) =-1.D0
999 RETURN
END

DEQR0001
DEQR0002
DEQR0003
DEQR0004
DEQR0005
DEQR0006
DEQR0007
DEQR0008
DEQR0009
DEQR0010
DEQR0011
DEQR0012
DEQR0013
DEQR0014
DEQR0015
DEQR0016
DEQR0017

Subroutine: DEQ3

Purpose: Integrates three second-order equations in double precision. (See also DEQD).

Calling Sequence: CALL DEQ3 (FSUB, ϕ SUB)

Input and Output

I/ ϕ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
	FSUB				Name of subroutine which evaluates derivatives
	ϕ SUB				Name of subroutine which controls output.

Common storages used: //(40), /DQ3COM/

Subroutines required: FSUB, ϕ SUB

DEQ3-1

Restrictions

An "EXTERNAL FSUB, ØSUB" statement must be included in the calling program. The subroutines FSUB, ØSUB must have zero length call lists. For use with OVERLAY, FSUB, ØSUB and the labelled common DQ3COM must be included in the link containing DEQ3 or in a higher link.

The functions of these subroutines are

FSUB Computes the second derivatives of the dependent variables, and any stopping functions.

ØSUB Provides for output at completion of integration of each interval (regular points) and at certain points obtained by interpolation.

The Kutta mode may not be used if any interpolation is required (i.e., even interval output or stopping functions).

DEQ3-2

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SIBFTC MC133E XR3,M94,NODD,LIST
SUBROUTINE DEQ3 (FSUB,OSUB)
DEQ3 - DOUBLE PRECISION INTEGRATION
OF EQUATIONS OF STATE
COMMON /DQ3COM/NDEQ(10),CDEQ(10),XDEQ (4)
1 ,ADEQ (8),RDEQ (8),VDEQ (8),FDEQ (8,10)
DOUBLE PRECISION ADEQ,FDEQ,RDEQ,VDEQ,XDEQ
1 ,X,XF,XE,XO
EQUIVALENCE (NDEQ( 1),NI),(CDEQ( 1),H),(XDEQ( 1),X)
1 ,(NDEQ( 2),NO),(CDEQ( 2),HP),(XDEQ( 2),XF)
2 ,(NDEQ( 3),NS),(CDEQ( 3),HO),(XDEQ( 3),XE)
3 ,(NDEQ( 4),NA),(CDEQ( 4),HU),(XDEQ( 4),XO)
4 ,(NDEQ( 5),NC),(CDEQ( 5),HL)
5 ,(NDEQ( 6),NE),(CDEQ( 6),HE)
6 ,(NDEQ( 7),NT),(CDEQ( 7),EU)
7 ,(NDEQ( 8),IP),(CDEQ( 8),EL)
8 ,(NDEQ( 9),IT)
COMMON STP(40)
DOUBLE PRECISION TMP(20)
EQUIVALENCE (STP(1),TMP)
DOUBLE PRECISION ALP(4),CQD(12),CQX(4),DCN(2),DX,FLT(7)
DIMENSION CAE (7),CDB(7,2),CQK (20),SFL (14)
1 ,CAI(6,6),CHV(6,3),IN (8),SLP (8)
2 ,CCV(8,6),CQA (12)
EQUIVALENCE (ALP(1),SLP),(FLT(1),SFL)
DATA CAE/ 1., -6., 15., -20., 15., -6., 1./
DATA CAI/5*0.,60., -12., 75., -200., 300., -300., 137.
1 , -50., 305., -780., 1070., -770., 225.
2 , -105., 615., -1470., 1770., -1065., 255.
3 , -120., 660., -1440., 1560., -840., 180.
4 , -60., 300., -600., 600., -300., 60./
DATA CCV/ 3864., -3200., -6496., -1539., 12327., -4300., -224., 324.
1 , 1176., -400., -3584., -972., 7812., -3600., 0., 324.
2 , 126., 300., -1344., -405., 3717., -1800., 0., 162.
3 , 0., 0., 0., 0., 756., 0., 0., 0.
4 , 84., -200., 224., 81., -357., 700., 224., 0.
5 , -336., 800., -896., -324., 1092., -800., 896., 324./
DATA CDB/ -224., -112., -48., 28., 210., 140., 7.
1 , -240., -1008., -560., 210., 2100., 1575., 84./
DATA CHV/ -0.01171875, 0.08203125, -0.2734375, 0.8203125
1 , 0.41015625, -0.02734375, 0.01171875, -0.09865625
2 , 0.5859375, 0.5859375, -0.09865625, 0.01171875
3 , 0.02734375, -0.17578125, 0.4921875, -0.8203125
4 , 1.23046875, 0.24609375/
DATA CQA/ -863., 5260., -13474., 18752., -15487., 10852.
1 , -3325., 20139., -51086., 69874., -55461., 29939./
DATA CQK/ 2268., 0., 2268., 1680., -196., 784., 420.
1 , 1400., 448., 0., 4536., -3969., 8505., 13944.
2 , -15680., 6272., 420., 2000., 1792., 324./
DATA CQX/ 1.0D+0, 1.0D+0, .75D+0, .30D+0/
DATA DCN/ 3.168648818567611, 3.966129870129870/
DATA FLT/ 1.0D, 2.0D, 3.0D, 4.0D, 5.0D, 6.0D, 7.0D/
C**** PROLOGUE SET FIXED DATA AND SWITCHES
100 DO 101 I=1,8
101 IN(I) = I
HP = 0.
EL = EU/128.
ASSIGN 234 TO LLA
ASSIGN 480 TO LLB
ASSIGN 480 TO LLC
ASSIGN 200 TO LLD
ASSIGN 521 TO LLE
ASSIGN 200 TO LLF
IF (NI.EQ.0) GO TO 103
IF (NS.NE.0) ASSIGN 502 TO LLE
IF (NA.NE.0) ASSIGN 450 TO LLC
ASSIGN 202 TO LLF
IF (NC.NE.2) ASSIGN 220 TO LLF
103 IRT = 1

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    IRS = 0
    IPT = 0
    IP = 2
    H = 0.
    ALP(4) = 0.D0
    GO TO 480
140 H = HO
    TMP(1) = XF-X
    IF (STP(1).LT.0.) H=-H
    IF (STP(1)/H.GT.2.) GO TO 141
    H = TMP(1)/2.D0
    IPT = 3
141 IP = 3
    IF (ABS(H).EQ.0.) GO TO 221
    IP = 0
    IT = 0
    NI = NE+1
    NT = NE+NS
C
C**** CONTROL          SELECT COMPUTATIONAL PATH
C
C    COMPUTE AND LOAD DERIVATIVES
200 I = IN(1)
    DO 201 J=1,7
201 IN(J) = IN(J+1)
    IN(8) = I
202 CALL FSUB
    I = IN(8)
    DO 203 J=1,NT
203 FDEQ(J,I) = ADEQ(J)
    GO TO (220,401,480,440) ,IRT
C
C    OUTPUT CONTROL
220 IF (IP.NE.0) GO TO 221
    IF (NO.EQ.0) GO TO 233
221 CALL OSUB
C
C    IDENTIFY POINT TYPE AND SELECT PATH
230 IF (IP) 231,232,232
231 IPT = 0
    NS = 0
    NT = NE
    XF = XF+FLT(7)*DX
    GO TO 300
232 IF (IP-1) 233,520,480
233 GO TO LLA, (234,400,300)
234 IF (HP.EQ.0.) GO TO 400
    ASSIGN 400 TO LLA
    IF (NI.EQ.0) GO TO 400
    ASSIGN 300 TO LLA
    ASSIGN 420 TO LLB
    ASSIGN 500 TO LLD
    GO TO 400
C
C**** NEXT          INTERVAL SIZE CONTROL
C
C    SELECT NEXT INTERVAL SIZE
300 IF (IDL.GT.0) GO TO 301
    IF (ABS(H).GE.HU) GO TO 404
    GO TO 340
301 IDL = IDL-1
    GO TO 404
C
C    DOUBLE SPACING OF ADAMS ORDINATES
340 H = H*SFL(3)
    I = IN(7)
    J = IN(5)
    IN(7) = IN(6)
    IN(6) = IN(4)
    IN(5) = IN(2)
    IN(4) = I
    IN(2) = J
    DO 342 I=1,NT
    TMP(1) = 0.D0
    TMP(2) = 0.D0

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DEQ30075
DEQ30076
DEQ30077
DEQ30078
DEQ30079
DEQ30080
DEQ30081
DEQ30082
DEQ30083
DEQ30084
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DEQ30086
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DEQ30142
DEQ30143
DEQ30144
DEQ30145
DEQ30146
DEQ30147
DEQ30148
DEQ30149

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DO 341 J=1,7
L = IN(J+1)
TMP(1) = TMP(1)+CDB(J,1)*FDEQ(I,L)
341 TMP(2) = TMP(2)+CDB(J,2)*FDEQ(I,L)
L = IN(4)
FDEQ(I,L) = TMP(1)
L = IN(3)
342 FDEQ(I,L) = TMP(2)
GO TO 403
C
C HALVE SPACING OF ADAMS ORDINATES
360 H = H/SFL(3)
DO 363 I=1,NT
DO 361 J=1,6
361 STP(J) = 0.
DO 362 J=1,6
L = IN(J+1)
TMP(4) = FDEQ(I,L)
DO 362 K=1,3
362 TMP(K) = TMP(K)+TMP(4)*CHV(J,K)
DO 363 J=1,3
L = IN(J+1)
363 FDEQ(I,L) = TMP(J)
I = IN(8)
IN(8) = IN(7)
IN(7) = IN(4)
IN(4) = IN(5)
IN(5) = IN(2)
IN(2) = I
GO TO 403
C
C**** QUADRATURE INTEGRATE TO X0+H
C
C LOCATE KUTTA SUBSTITUTION COEFFICIENTS
400 IRT = 2
ISB = 9
ICN = 1
401 ISB = ISB-1
DX = H*CQX(ISB-4)
DO 402 I=ISB,8
CQD(I-2) = CQK(ICN)/4536.D0
CQD(I+4) = CQK(ICN+10)/4536.D0
402 ICN = ICN+1
GO TO 410
403 IDL = 2
DX = H
ALP(2) = 60.D0/DX
ALP(3) = ALP(2)/DX
404 IRT = NRT
C
C INTEGRATE/PREDICT
410 X = X0+DX
DO 413 I=1,3
TMP(1) = 0.D0
TMP(2) = 0.D0
DO 411 J=ISB,8
L = IN(J)
TMP(1) = TMP(1)+CQD(J-2)*FDEQ(I,L)
411 TMP(2) = TMP(2)+CQD(J+4)*FDEQ(I,L)
DO 412 J=1,2
412 TMP(J) = FDEQ(I,10)+DX*TMP(J)
RDEQ(I) = FDEQ(I,9)+DX*TMP(1)
413 VDEQ(I) = TMP(2)
IF (IRT.NE.2) GO TO 200
IF (ISB.GT.5) GO TO 200
GO TO LLB, (480,420,440,432)
C
C**** CONVERT CHANGE FROM START TO ADAMS MODE
C
C COMPUTE ADAMS ORDINATES
420 H = H/SFL(3)
DO 422 I=1,NT
DO 421 J=1,6
TMP(J) = 0.D0
DO 421 L=1,8
DEQ30150
DEQ30151
DEQ30152
DEQ30153
DEQ30154
DEQ30155
DEQ30156
DEQ30157
DEQ30158
DEQ30159
DEQ30160
DEQ30161
DEQ30162
DEQ30163
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DEQ30209
DEQ30210
DEQ30211
DEQ30212
DEQ30213
DEQ30214
DEQ30215
DEQ30216
DEQ30217
DEQ30218
DEQ30219
DEQ30220
DEQ30221
DEQ30222
DEQ30223
DEQ30224

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421	TMP(J) = TMP(J)+CCV(L,J)*FDEQ(I,L)	DEQ30225
	DO 422 J=3,8	DEQ30226
422	FDEQ(I,J) = TMP(J-2)/756.DO	DEQ30227
C		DEQ30228
	SET ADAMS INTEGRATION PARAMETERS	DEQ30229
430	ISB = 3	DEQ30230
	NRT = 3	DEQ30231
	ASSIGN 480 TO LLB	DEQ30232
	DO 431 I=1,12	DEQ30233
431	CQD(I) = CQA(I)/10080.DO	DEQ30234
	IF (NA+NC.EQ.0) GO TO 480	DEQ30235
	NRT = 4	DEQ30236
	IDL = 4	DEQ30237
	IP = IPT	DEQ30238
	ASSIGN 432 TO LLB	DEQ30239
	IF (IP.NE.0) GO TO 221	DEQ30240
	GO TO 480	DEQ30241
432	IF (IDL.GT.1) GO TO 480	DEQ30242
	ASSIGN 440 TO LLB	DEQ30243
C		DEQ30244
C****	CORRECT TEST ACCURACY AND ADD ADAMS CORRECTOR	DEQ30245
C		DEQ30246
C	ADD CORRECTOR AND COMPUTE MAXIMUM	DEQ30247
440	STP(1) = 0.	DEQ30248
	DO 442 I=1,3	DEQ30249
	STP(2) = ABS(VDEQ(I))	DEQ30250
	IF (SFL(1).GT.STP(2)) STP(2) = SFL(1)	DEQ30251
	TMP(3) = 0.DO	DEQ30252
	DO 441 J=1,7	DEQ30253
	L = IN(J+1)	DEQ30254
441	TMP(3) = TMP(3)+CAE(J)*FDEQ(I,L)	DEQ30255
	TMP(3) = DX*TMP(3)/DCN(1)	DEQ30256
	TMP(2) = DX*TMP(3)/DCN(2)	DEQ30257
	STP(2) = ABS(STP(5))/STP(2)	DEQ30258
	IF (STP(2).GT.STP(1)) STP(1) = STP(2)	DEQ30259
	IF (NC.EQ.0) GO TO 442	DEQ30260
	RDEQ(I) = RDEQ(I)+TMP(2)	DEQ30261
	VDEQ(I) = VDEQ(I)+TMP(3)	DEQ30262
442	CONTINUE	DEQ30263
	GO TO LLC, (480,450)	DEQ30264
C		DEQ30265
C	TEST ACCURACY	DEQ30266
450	IF (STP(1).LT.EL) GO TO 480	DEQ30267
	IDL = IDL+1	DEQ30268
	IF (STP(1).LT.EU) GO TO 480	DEQ30269
	IF (ABS(H).LE.HL) GO TO 480	DEQ30270
	GO TO 360	DEQ30271
C		DEQ30272
C****	ADVANCE STORE ICS FOR NEXT INTERVAL	DEQ30273
C		DEQ30274
480	IRT = 1	DEQ30275
	HP = H	DEQ30276
	DO 481 I=1,3	DEQ30277
	FDEQ(I,9) = RDEQ(I)	DEQ30278
481	FDEQ(I,10) = VDEQ(I)	DEQ30279
	X0 = X	DEQ30280
	IF (IP-2) 482,140,999	DEQ30281
482	GO TO LLD, (500,200)	DEQ30282
C		DEQ30283
C****	TEST LOCATE AND ORDER SPECIAL POINTS	DEQ30284
C		DEQ30285
C	FIND STOPPING POINTS	DEQ30286
500	ALP(4) = 0.DO	DEQ30287
	IPT = 0	DEQ30288
	ITT = 0	DEQ30289
	ALP(1) = (XF-X)/DX	DEQ30290
	IF (SLP(1).GT.0.) GO TO 501	DEQ30291
	ALP(4) = ALP(1)	DEQ30292
	IPT = 3	DEQ30293
501	GO TO LLE, (502,521)	DEQ30294
502	K = IN(8)	DEQ30295
	L = IN(7)	DEQ30296
	DO 512 I=N1,NT	DEQ30297
	TMP(2) = FDEQ(I,L)-FDEQ(I,K)	DEQ30298
	M = 3	DEQ30299

503	GO TO 504	DEQ30300
	K = IN(6)	DEQ30301
	M = 2	DEQ30302
	TMP(2) = (TMP(2)+FDEQ(I,K)-FDEQ(I,L))/FLT(2)	DEQ30303
504	IF (STP(3).EQ.0.) GO TO 506	DEQ30304
	TMP(1) = FDEQ(I,L)/TMP(2)-FLT(1)	DEQ30305
505	IF (STP(1).GT.SLP(7)) GO TO 506	DEQ30306
	IF (STP(1)+SFL(3).LE.0.) GO TO 506	DEQ30307
	GO TO (512,507,507) ,M	DEQ30308
506	GO TO (512,512,503) ,M	DEQ30309
507	M = 1	DEQ30310
	DO 508 J=1,12	DEQ30311
508	STP(J+6) = 0.	DEQ30312
	DO 509 J=1,6	DEQ30313
	L = IN(J+2)	DEQ30314
	TMP(10) = FDEQ(I,L)	DEQ30315
	DO 509 K=1,6	DEQ30316
509	TMP(K+3) = TMP(K+3)+TMP(10)*CAI(J,K)	DEQ30317
	DO 510 J=1,5	DEQ30318
	K = 5-J	DEQ30319
510	TMP(9) = TMP(K+4)+TMP(9)*TMP(1)/FLT(K+1)	DEQ30320
	TMP(3) = TMP(9)/TMP(2)/60.DO	DEQ30321
	TMP(1) = TMP(1)+TMP(3)	DEQ30322
	IF (ABS(STP(5)).GT..1E-11) GO TO 505	DEQ30323
	IF (TMP(1)-ALP(4)) 511,511,512	DEQ30324
511	IPT = 3	DEQ30325
	ITT = I	DEQ30326
	ALP(4) = TMP(1)	DEQ30327
512	CONTINUE	DEQ30328
	GO TO 521	DEQ30329
C		DEQ30330
C	FIND EVEN-INTERVAL POINT	DEQ30331
520	IRS = 1	DEQ30332
521	ALP(1) = (XE-X)/DX	DEQ30333
	IF (ALP(1)-ALP(4)) 523,522,524	DEQ30334
522	IF (IPT.NE.0) GO TO 524	DEQ30335
523	IP = 1	DEQ30336
	XE = XE+HE	DEQ30337
	GO TO 525	DEQ30338
524	IP = IPT	DEQ30339
	IT = ITT	DEQ30340
	ALP(1) = ALP(4)	DEQ30341
525	IF (SLP(1).NE.0.) GO TO 610	DEQ30342
	IF (IPT.NE.0) GO TO 221	DEQ30343
	IF (IRS) 600,600,540	DEQ30344
C		DEQ30345
C****	RESET RESTORE INTERVAL END-POINT	DEQ30346
C		DEQ30347
540	IRS = 0	DEQ30348
	X = X0	DEQ30349
	L = IN(8)	DEQ30350
	DO 541 I=1,3	DEQ30351
	RDEQ(I) = FDEQ(I, 9)	DEQ30352
541	VDEQ(I) = FDEQ(I,10)	DEQ30353
	DO 542 I=1,NT	DEQ30354
542	ADEQ(I) = FDEQ(I,L)	DEQ30355
C		DEQ30356
C****	INTERPOLATE COMPUTE INTEGRALS AT X0+T*H	DEQ30357
C		DEQ30358
600	IF (SLP(1).EQ.0.) GO TO LLF, (200,202,220)	DEQ30359
C		DEQ30360
C	COMPUTE INTERPOLATION COEFFICIENTS	DEQ30361
610	TMP(1) = ALP(1)	DEQ30362
	DO 611 I=1,6	DEQ30363
611	TMP(I+1) = TMP(1)*TMP(I)/FLT(I+1)	DEQ30364
	DO 612 I=17,40	DEQ30365
612	STP(I) = 0.	DEQ30366
	DO 614 I=1,6	DEQ30367
	DO 613 J=1,6	DEQ30368
	TMP(8) = CAI(I,J)	DEQ30369
	TMP(I+ 8) = TMP(I+ 8)+TMP(8)*TMP(J)	DEQ30370
613	TMP(I+14) = TMP(I+14)+TMP(8)*TMP(J+1)	DEQ30371
	TMP(I+ 8) = TMP(I+ 8)/ALP(2)	DEQ30372
614	TMP(I+14) = TMP(I+14)/ALP(3)	DEQ30373
C		DEQ30374

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C   INTERPOLATE TO X0+T*H
620 TMP(6) = DX*TMP(1)
    X = X0+TMP(6)
    DO 621 I=1,3
      VDEQ(I) = FDEQ(I,10)
      RDEQ(I) = FDEQ(I, 9)+DX*FDEQ(I,10)
    DO 621 J=1,6
      L = IN(J+2)
      VDEQ(I) = VDEQ(I)+FDEQ(I,L)*TMP(J+ 8)
621 RDEQ(I) = RDEQ(I)+FDEQ(I,L)*TMP(J+14)
    GO TO 221
C
C**** RETURN          STOP POINT REACHED
C
999 RETURN
END

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DEQ30375
DEQ30376
DEQ30377
DEQ30378
DEQ30379
DEQ30380
DEQ30381
DEQ30382
DEQ30383
DEQ30384
DEQ30385
DEQ30386
DEQ30387
DEQ30388
DEQ30389

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Subroutine: DGTRN

Purpose: To compute the transformation matrix corresponding to:
(1) a rotation about a given line, or (2) a sequence of rotations about the coordinate axes.

Calling Sequence: CALL DGTRN (T, N, A, M)
CALL DGTRN (T, O, A, W)

Input and Output

I/ø	Symbolic Name of Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	N	(M)			Axes of rotation.
I	A	d(M)			Angles of rotation.
I	M				Number of rotations.
I	W	d(3)			Vector along line of rotation.
ø	T	d(3, 3)			Transformation matrix.

Common storages used: 72 cells

Subroutines required: DGTSN

DGTRN-1

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$IBFTC MC132L XR3,M94,NODD,LIST
SUBROUTINE DGTRN (T,N,A,M)
DOUBLE PRECISION A(10),CA,SA,DSIN,DCOS
COMMON SAV(32),CA(10),SA(10)
1 MG = 1
IF (N.NE.0) MG=M
DO 2 I=1,MG
CA(I) = DCOS(A(I))
2 SA(I) = DSIN(A(I))
CALL DGTSN (T,N,SA,CA,M)
RETURN
END
```

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DGTR0001
DGTR0002
DGTR0003
DGTR0004
DGTR0005
DGTR0006
DGTR0007
DGTR0008
DGTR0009
DGTR0010
```

Subroutine: DGTSN

Purpose: To compute the transformation matrix corresponding to: (1) a rotation about a given line, or (2) a sequence of rotations about the coordinate axes.

Calling Sequences: CALL DGTSN (T,N,SA,CA,M)
CALL DGTSN (T,O,SA,CA,W)

Input and Output

I/∅	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	N	(M)			Axes of rotations.
I	SA	d(M)			Sines of angles of rotation.
I	CA	d(M)			Cosines of angles of rotation.
I	M				Number of rotations.
I	W	d(3)			Vector along line of rotation.
∅	T	d(3,3)			Transformation matrix.

Common storages used: 32 cells

Subroutines required: DVNORM

DGTSN-1

SIBFTC MC132N XR3,M94,NODD,LIST	
SUBROUTINE DGTSN (T,N,SA,CA,M)	DGTS0001
DOUBLE PRECISION CA(10),SA(10),T(9)	DGTS0002
1 ,ST(9),WT(5),S,C	DGTS0003
INTEGER N(10),NN(4)	DGTS0004
COMMON SS(18),WT,S,C	DGTS0005
EQUIVALENCE (SS(1),ST)	DGTS0006
DATA NN/6,7,2,3/	DGTS0007
C	DGTS0008
1 MG = 1	DGTS0009
ASSIGN 20 TO LL	DGTS0010
IF (N.EQ.0) GO TO 10	DGTS0011
DO 2 I=1,18	DGTS0012
2 SS(I) = 0.	DGTS0013
DO 3 I=1,18,8	DGTS0014
3 SS(I) = 1.	DGTS0015
MG = M	DGTS0016
ASSIGN 30 TO LL	DGTS0017
C	DGTS0018
10 DO 35 I=1,MG	DGTS0019
C = CA(I)	DGTS0020
S = SA(I)	DGTS0021
GO TO LL, (20,30)	DGTS0022
C	DGTS0023
20 CALL DVNORM(M,WT)	DGTS0024
L = 1	DGTS0025
WT(4) = 1.-C	DGTS0026
DO 21 J=1,3	DGTS0027
WT(5) = WT(4)*WT(J)	DGTS0028
DO 21 K=1,3	DGTS0029
ST(L) = WT(5)*WT(K)	DGTS0030
21 L = L+1	DGTS0031
DO 22 J=1,9,4	DGTS0032
22 ST(J) = ST(J)+C	DGTS0033
DO 23 J=1,3	DGTS0034
K = NN(J)	DGTS0035
L = NN(J+1)+1	DGTS0036
ST(K) = ST(K)-S*WT(J)	DGTS0037
23 ST(L) = ST(L)+S*WT(J)	DGTS0038
GO TO 35	DGTS0039
C	DGTS0040
30 J = IABS(N(I))+1	DGTS0041
IF (N(I).LT.0) S=-S	DGTS0042
IF (J.GT.3) J=1	DGTS0043
K = J+1	DGTS0044
IF (K.GT.3) K=1	DGTS0045
DO 31 L=1,3	DGTS0046
WT(4) = ST(J)	DGTS0047
ST(J) = C*WT(4)+S*ST(K)	DGTS0048
ST(K) = -S*WT(4)+C*ST(K)	DGTS0049
J = J+3	DGTS0050
31 K = K+3	DGTS0051
35 CONTINUE	DGTS0052
DO 36 I=1,9	DGTS0053
36 T(I) = ST(I)	DGTS0054
RETURN	DGTS0055
END	

Subroutine: DIFCØR

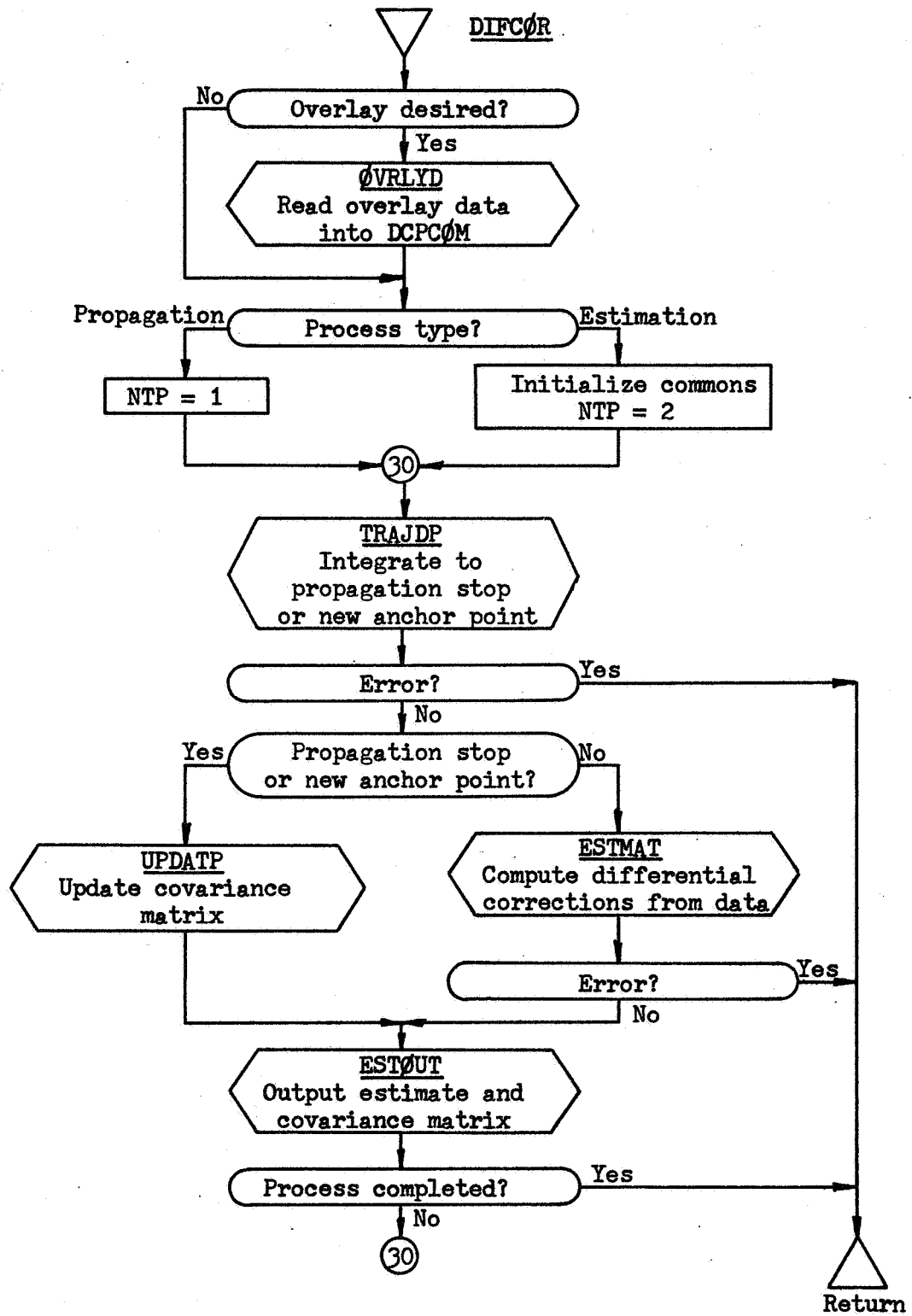
Purpose: To control integration of the nominal trajectory and the accumulation of differential corrections to the estimate.

Calling Sequence: CALL DIFCØR

Common storages used: /DCPCØM/, /DCRCØM/, /SBFCØM/

Subroutines required: ESTMAT, ESTØUT, ØVRLYD, TRAJDP, UPDATP

DIFCØR-1



DIFCØR-2

```

$IBFTC MC133H XR3,M94,NODD,LIST
SUBROUTINE DIFCOR
C CONTROL ROUTINE FOR DCP ESTIMATION/PROPAGATION LINKS
C
COMMON /DCPCOM/CDCP(900)
DIMENSION STIMNX(20)
DOUBLE PRECISION STIMR(2)
EQUIVALENCE (CDCP(111),IERR ),(CDCP(740),NPREST)
1 ,(CDCP(758),INDSTA) ,(CDCP(732),NPROVR)
2 ,(CDCP(756),IPROCS) ,(CDCP(123),STIMNX)
3 ,(CDCP(105),MAXZ ) ,(CDCP(781),STIMR )
C
COMMON /SBFCOM/CSBF(12),RBF(45),VBF(45,8)
DOUBLE PRECISION RBF,VBF
EQUIVALENCE (CSBF(2),NTP)
C
COMMON /DCRCOM/ZBAR(30),ZHAT(30),ZTOT(30),DCSAVE(74)
DOUBLE PRECISION ZBAR ,ZHAT ,ZTOT
C
LOGICAL LTRAJ
C
601 FORMAT(1H0//45X,24H*** PROPAGATION LINK ***/1H )
602 FORMAT(1H0//45X,23H*** ESTIMATION LINK ***/1H )
603 FORMAT(1H0//43X,29H*** EXIT PROPAGATION LINK ***)
604 FORMAT(1H0//43X,28H*** EXIT ESTIMATION LINK ***)
C
C READ OVERLAY DATA
1 IF (NPROVR.NE.0) CALL OVRLYD(CDCP)
GO TO (10,20) ,IPROCS
C
C PROPAGATE
10 NTP = 1
WRITE (6,601)
GO TO 30
C
C ESTIMATE
20 CONTINUE
DO 21 I=1,MAXZ
ZBAR(I) = 0.D0
ZHAT(I) = 0.D0
21 ZTOT(I) = 0.D0
DO 22 I=1,6
22 CSBF(I) = 0.
NTP = 2
LTRAJ = .FALSE.
WRITE (6,602)
C
C INTEGRATE
30 CALL TRAJDP
IF (IERR.NE.0) GO TO 999
IF (NTP.GE.3) GO TO 33
31 CALL UPDATP
NPREST = 0
32 CALL ESTOUT
IF (NTP.EQ.1) GO TO 950
IF (NTP.EQ.4) GO TO 900
NTP = 2
GO TO 30
33 CALL ESTMAT (LTRAJ)
IF (IERR.NE.0) GO TO 999
GO TO 32
C
900 STIMNX(INDSTA) = STIMR(2)
WRITE (6,604)
GO TO 999
950 WRITE (6,603)
999 RETURN
END
DIFC0001
DIFC0002
DIFC0003
DIFC0004
DIFC0005
DIFC0006
DIFC0007
DIFC0008
DIFC0009
DIFC0010
DIFC0011
DIFC0012
DIFC0013
DIFC0014
DIFC0015
DIFC0016
DIFC0017
DIFC0018
DIFC0019
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DIFC0055
DIFC0056
DIFC0057
DIFC0058
DIFC0059
DIFC0060
DIFC0061
DIFC0062
DIFC0063
DIFC0064
DIFC0065
DIFC0066

```

Subroutine: DINVRT

Purpose: To invert a double precision matrix of any dimension,
possibly stored as a submatrix of a larger array.
Replaces the matrix by its inverse.

Calling Sequence: CALL DINVRT(A,N,M)

Input and Output:

I/ ϕ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I ϕ	A	d(M, l)			I = matrix to be inverted. ϕ = inverse of input matrix.
I	N				Dimension of matrix to be inverted.
I	M				Dimension of storage array.

Common storages used: 7 + N

Subroutines required: None

DINVRT-1

Method:

Reference: F.B. Hildebrand, "Introduction to Numerical Analysis,"
McGraw-Hill, 1956.

The inverse of the $n \times n$ matrix $A = \{A_{ij}\}$ is the solution, $X = A^{-1}$, of
the matrix equation

$$AX = I \quad (1)$$

It is obtained by premultiplying both sides of (1) by a sequence of
matrices chosen to eliminate the matrix from the left-hand side.

The sequence of matrices used in this subroutine constitute a Gauss-
Jordan reduction. Storage is shared between the reduced matrix and its
inverse during the inversion.

We define an augmented matrix,

$$M = (A : I) \quad (2)$$

We may place 1 on the main diagonal and null the remaining elements
of the first column by premultiplying by the matrix

$$E_1 = \left[\begin{array}{c|c} 1/a_{11} & \bar{0} \\ \hline -a_{21}/a_{11} & \\ -a_{31}/a_{11} & \\ \vdots & \\ \vdots & \\ \vdots & \\ -a_{n1}/a_{11} & I \end{array} \right] \quad (3)$$

DINVRT-2

Similarly, the second column diagonal element is replaced by 1 and the remaining elements of the column are nulled by premultiplication by

$$E_2 = \begin{bmatrix} 1 & -a'_{12}/a'_{22} & 0 \\ 0 & 1/a'_{22} & 0 \\ & -a'_{32}/a'_{22} & \\ & \vdots & \\ \bar{0} & -a'_{n2}/a'_{22} & I \end{bmatrix} \quad (4)$$

where primes denote the elements resulting from the previous premultiplication.

We proceed columnwise across A in this fashion, obtaining finally an auxiliary matrix,

$$M' = C' (A \vdots I) = (A' \vdots C) \quad (5)$$

$$C' = E_n E_{n-1} \cdots E_2 E_1$$

The determinant of the matrix A is the product of the divisors of the E_j , as may be seen from

$$| E_j | = 1/a'_{jj} \quad (6)$$

$$1 = | I | = | E_n | | E_{n-1} | \cdots | E_1 | | A |$$

The principal disadvantage of this process as described is the use of $2n^2$ storage locations for the augmented matrix. The process is however, easily modified to share storage between the reduced matrix A' and the uncompleted inverse.

DINVRT-3

Since the columns preceding the j^{th} have non-zero elements only on the main diagonal before the j^{th} premultiplication, they are not changed during that premultiplication. Similarly, the columns after the j^{th} in the right half of the augmented matrix are not changed. We need not, then, store the columns known to contain 0's and 1's and both right and left matrices may be stored in the single $n \times n$ array, with appropriate attention to the j^{th} row.

We obtain the process

$$a'_{ij} = \begin{cases} 1/a'_{jj} & i = j \\ -a'_{ij} & i \neq j \end{cases} \quad (7)$$

$$a'_{ik} = a'_{ik} + a'_{jj} a'_{jk} \quad i, k \neq j$$

$$a'_{ji} = a'_{ji} \cdot a'_{jj} \quad i \neq j$$

repeated for $j = 1, 2, \dots, n$. Again, after n steps, $X = A^{-1}$ occupies the spaces originally occupied by A .

Finally, we note that maximum numerical accuracy is obtained if the diagonal element a'_{ji} has the largest magnitude of the elements remaining in the unused rows and columns in A' . Then, if on the j^{th} step, a'_{ik} is the element of largest magnitude, we may eliminate the k^{th} column by placing a 1 in a'_{ik} by the process

$$a'_{lk} = \begin{cases} 1/a'_{ik} & l = i \\ -a'_{lm}/a'_{ik} & l \neq i \end{cases} \quad (8)$$

$$a'_{lm} = a'_{lm} + a'_{lk} a'_{ik} \quad l \neq i, m \neq k$$

$$a'_{il} = a'_{il} \quad l \neq k$$

DINVRT-4

Although the column available for storage is the k^{th} , the column of the right matrix stored there is its i^{th} . In addition, the resulting left half is a matrix with ones on each row and column, but not necessarily on the diagonal. It is necessary to record the locations of the 1's and to use this record to reorder the rows and columns of the inverse.

DINVRT-5

SIBFTC MC13MH XR3,M94,NODD,LIST	
SUBROUTINE DINVRT(A,N,M)	DINV0001
REPLACES THE N BY N MATRIX A BY ITS INVERSE.	DINV0002
A IS STORED IN THE CALLING PROGRAM AS A SUBMATRIX OF AN M BY M	DINV0003
ARRAY.	DINV0004
C	DINV0005
C	DINV0006
C	DINV0007
DOUBLE PRECISION A,T,S	DINV0008
DIMENSION A(1)	DINV0009
COMMON T,S,ND,NM,NI,IN(20)	DINV0010
1 CONTINUE	DINV0011
NI=N	DINV0012
NM=M	DINV0013
IF(NI)999,999,2	DINV0014
2 ND=NM-NI	DINV0015
NN=NI*NM	DINV0016
DO 3 J=1,NI	DINV0017
3 IN(J)=0	DINV0018
10 DO 29 L=1,NI	DINV0019
T=0.0	DINV0020
K=1	DINV0021
DO 16 J=1,NI	DINV0022
DO 12 I=1,NI	DINV0023
IF(IN(I)-J)12,11,12	DINV0024
11 K=K+NM	DINV0025
GO TO 16	DINV0026
12 CONTINUE	DINV0027
DO 15 I=1,NI	DINV0028
IF(IN(I))15,13,15	DINV0029
13 IF(DABS(A(K))-DABS(T)) 15,15,14	DINV0030
14 ICOL=J	DINV0031
IROW=I	DINV0032
K1=K	DINV0033
T=A(K)	DINV0034
15 K=K+1	DINV0035
K=K+ND	DINV0036
16 CONTINUE	DINV0037
20 IN(IROW)=ICOL	DINV0038
A(K1)=1.0	DINV0039
21 K=K1-IROW	DINV0040
DO 24 I=1,NI	DINV0041
K=K+1	DINV0042
IF(IROW-I)22,24,22	DINV0043
22 S=-A(K)/T	DINV0044
A(K)=0.0	DINV0045
J1=I	DINV0046
DO 23 J=IROW,NN,NM	DINV0047
A(J1)=A(J1)+S*A(J)	DINV0048
23 J1=J1+NM	DINV0049
24 CONTINUE	DINV0050
DO 25 J=IROW,NN,NM	DINV0051
25 A(J)=A(J)/T	DINV0052
29 CONTINUE	DINV0053
30 DO 39 I=1,NI	DINV0054
DO 31 J=I,NI	DINV0055
IF(IN(J)-I)31,32,31	DINV0056
31 CONTINUE	DINV0057
32 IF(J-I)33,39,33	DINV0058
33 CONTINUE	DINV0059
IN(J)=IN(I)	DINV0060
J1 = (IN(I)-1)*NM	DINV0061
K1=(I-1)*NM	DINV0062
DO 34 K=I,NN,NM	DINV0063
S=A(K)	DINV0064
A(K)=A(J)	DINV0065
A(J)=S	DINV0066
34 J=J+NM	DINV0067
DO 35 K=1,NI	DINV0068
K1=K1+1	DINV0069
J1=J1+1	DINV0070
S=A(K1)	DINV0071
A(K1)=A(J1)	DINV0072
35 A(J1)=S	DINV0073
39 CONTINUE	DINV0074
999 CONTINUE	
RETURN	
END	

Subroutine: DLUNE

Purpose: To compute the transformation from Earth's true equator equinox of date to Moon-fixed coordinates and the Moon's angular velocity.

Calling Sequence: CALL DLUNE (T,E,DE,DL,TR, ϕ M)

Input and Output

I/ ϕ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	T	d	t_E	sec	Time from 1950 Jan 0.0 ET
I	E	d	\bar{e}	rad	Mean obliquity of the ecliptic
I	DE	d	δe	rad	Nutation in obliquity
I	DL	d	δL	rad	Nutation in longitude
ϕ	TR	d(3,3)	T_{D2B}		Transformation matrix
ϕ	ϕ M	d	ω_M	rad/sec	Moon's angular velocity

Common storages used: 72 cells

Subroutines required: DGTRN

DLUNE-1

Notation

Symbol	Quantity
L	Longitude of the Sun, from the equinox of date.
Γ	Longitude of perigee of the Sun.
ϵ	Obliquity of the ecliptic.
L'	Longitude of the Moon, measured in the ecliptic from the mean equinox of date to the mean ascending node of the lunar orbit, and then along the orbit.
Γ'	Longitude of perigee of the Moon, measured as L' .
Ω	Longitude of the ascending node of the Moon's orbit on the ecliptic.
I	Inclination of the Moon's equator with the ecliptic.
φ'	Right ascension of the Moon's prime meridian, from the ascending node of the equator on the ecliptic.
$\bar{(\)}$	Mean ()
$\delta(\)$	Nutation or libration in ().
ω_M	Moon's angular velocity about its polar axis.

DLUNE-2

Method

The transformation from Earth's equator, equinox of date to Moon-fixed coordinates, T_{D2B} , may be written

$$T_{D2B} = T_3 (\varphi') T_1 (I) T_3 (\Omega + \delta L - 180) T_1 (\epsilon)$$

The nutations δL , $\delta \epsilon$ and the mean obliquity $\bar{\epsilon}$ are input quantities.

The librations δI , δL , $\delta \Omega$ are

$$\delta I = 0.0297222 \cos g_2 - 0.00305555 \cos g_3 + 0.0102777 \cos g_4$$

$$\delta L = 0.0163888 \sin g_1 - 0.003333 \sin g_2 + 0.005 \sin g_3$$

$$\delta \Omega = \csc \bar{I} \{ -0.0302777 \sin g_2 - 0.00305555 \sin g_3 + 0.0102777 \sin g_4 \}$$

where

$$g_1 = \bar{L} - \bar{\Gamma}$$

$$g_2 = \bar{L}' - \bar{\Gamma}'$$

$$g_3 = 2\bar{\Gamma}' - 2\bar{\Omega}$$

$$g_4 = g_2 + g_3 = \bar{L}' + \bar{\Gamma}' - 2\bar{\Omega}$$

The right ascension φ' is

$$\varphi' = L' - \Omega = (\bar{L}' - \bar{\Omega}) + \delta L' - \delta \Omega$$

and the polar component of angular velocity is

$$\begin{aligned} \omega_M &= L' - \Omega + \Omega \cos I \\ &= L' - \Omega (1 - \cos I) \end{aligned}$$

DLUNE-3

From reference 3, pp. 98, 107, the mean ephemerides of the Sun and Moon, reduced to the epoch T_{E1} , yield

$$\bar{I} = 1.535$$

$$\bar{\Omega} = 12.1127905556 - 1934.1440877778T_E + 0.002081T_E^2 + 0.22 \times 10^{-5}T_E^3$$

$$g_1 = 1.9993186111 + 129596578.550T_E - 0.559T_E^2 - 0.012T_E^3$$

$$g_2 = -144.4685381556 + 477198.6450208333T_E + 0.0092136T_E^2 + 1.439 \times 10^{-5}T_E^3$$

$$g_3 = 33.4623977778 + 12006.7621433333T_E - 0.02485T_E^2 - 2.94 \times 10^{-5}T_E^3$$

$$g_4 = g_2 + g_3$$

$$\bar{\varpi}' = g_2 + \frac{1}{2}g_3$$

The Moon's angular velocity is computed from

$$\bar{L}' = .2661702799661347 \times 10^{-5} - .12505386 \times 10^{-13}T_E$$

$$\bar{\Omega} = -.1069700561503563 \times 10^{-7} + .23018226 \times 10^{-13}T_E$$

$$\delta L' = .569493743 \times 10^{-10} \cos g_1 - .53527184 \times 10^{-9} \cos g_2 + .579490975 \times 10^{-11} \cos g_3$$

$$\delta \Omega = -.520641836 \times 10^{-7} \cos g_2 - .132200251 \times 10^{-9} \cos g_3 + .188410597 \times 10^{-7} \cos g_4$$

DLUNE-4

```

SIBFTC MC1320 XR3,M94,NODD,LIST
SUBROUTINE DLUNE (T,E,DE,DL,TR,OM)
DOUBLE PRECISION DE,DL,E,OM,T,TR
1          ,A(6,4),B(15),C,DSIN,DCOS
INTEGER    N(4)
COMMON    C(36)
DATA A/-.58 D-7,0.D0, .251 D-7,0.D0,-.513 D-7, .38 D-7
1          ,-.271 D-5,0.D0, .1608D-4,0.D0,-.4337D-4, .3632D-4
2          , .62830194306026630D+03,-.12505386          D-13
3          , .83286875416691855D+04, .23018226          D-13
4          , .20955753104977409D+03,-.33757182439704825D+02
5          ,-.34894692560095          D-01, .2661702799661347 D-05
6          ,-.2521451656358276 D+01,-.1069700561503563 D-07
7          , .584029016834644          ,-.3353000730389487 D+01 /
DATA B/ .569493743D-10, .28603852 D-03,-.520641836D-07
1          ,-.19727246 D-01,-.53329408 D-04,-.153527184D-09
2          ,-.58171824 D-04,-.132200251D-09,-.19908245 D-02
3          , .17937970 D-03, .579490975D-11, .87266463 D-04
4          , .188410597D-07, .66963710 D-02,-.51875025 D-03 /
DATA N/-3,-1,-3,-1/
C
1 C(1) = T/3.15576D+09
DO 2 I=1,6
2 C(I+1) = A(I,1)
DO 3 I=1,3
DO 3 J=1,6
3 C(J+1) = A(J,I+1)+C(1)*C(J+1)
C(8) = C(4)+C(6)
DO 4 I=1,7,2
C(1) = C(I+1)
C(I+ 8) = DCOS(C(1))
4 C(I+12) = DSIN(C(1))
C(17) = C(11)
DO 5 I=1,5
5 C(I+17) = B(I)*C(I+8)+B(I+5)*C(I+10)+B(I+10)*C(I+12)
C(6) = C(4)+C(19)-C(21)+C(6)/2.D0
C(3) = C(3)+C(18)
C(5) = C(5)+C(20)
C(8) = C(7)+C(21)+DL
C(9) = E+DE
C(7) = .026790804018112958+C(22)
OM = C(3)-C(5)+C(5)*DCOS(C(7))
CALL DGTRN (TR,N,C(6),4)
999 RETURN
END
DLUN0001
DLUN0002
DLUN0003
DLUN0004
DLUN0005
DLUN0006
DLUN0007
DLUN0008
DLUN0009
DLUN0010
DLUN0011
DLUN0012
DLUN0013
DLUN0014
DLUN0015
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DLUN0027
DLUN0028
DLUN0029
DLUN0030
DLUN0031
DLUN0032
DLUN0034
DLUN0035
DLUN0036
DLUN0037
DLUN0038
DLUN0039
DLUN0040
DLUN0041
DLUN0042
DLUN0043
DLUN0044

```

Subroutine: DMPLY

Purpose: To form the product of any two double precision matrices for which the product is defined.

Calling Sequence: CALL DMPLY(A,B,C,L,M,N,NA,NB,NC,IP)

Input and Output

I/ ϕ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	A	d(NA, l)			Pre-matrix.
I	B	d(NB, m)			Post-matrix.
ϕ	C	d(NC, n)			Product matrix.
I	L,M,N				Product dimensions.
I	NA,NB,NC*				Number of rows in storage arrays, A,B,C.
I	IP				Option index.

Common storages used: None

Subroutines required: None

*Restrictions. The number of rows of storage (NA,NB,NC) must be not less than the number of rows used in the product (L,M,N, respectively).

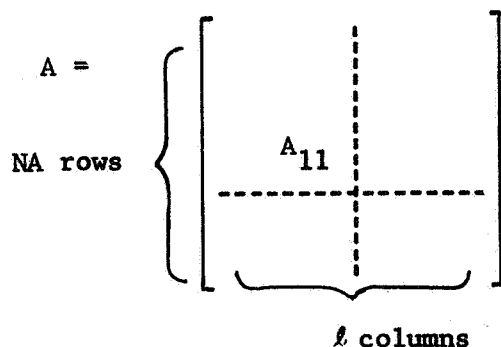
DMPLY-1

The subroutine allows multiplication of matrices stored as submatrices of large arrays. This allows operation with variable-dimensioned matrices within fixed dimensioned programs, and operations with partitioned matrices.

The options index, IP, allows transposing either or both of the factor matrices before multiplication. The subroutine output is

IP = 0: C = AB
 IP = 1: C = A^TB
 IP = 2: C = AB^T
 IP = 3: C = A^TB^T

The storage arrays are assumed to be partitioned in the form



and C_{11} is the product of A_{11} and B_{11} . Other submatrices in the arrays are ignored. The product dimensions are $L \times N$, with M terms in each summation. That is, the pre-matrix, A_{11} or its transpose, is $L \times M$, and the post-matrix B_{11} or its transpose, is $M \times N$.

For example, the call

CALL MTMPY (A,B,C,5,3,2,7,5,6,0)

DMPY-2

forms the product $C_{11} = A_{11} B_{11}$, where A_{11} , B_{11} , C_{11} are 5×3 , 3×2 ,
 5×2 matrices stored in $7 \times l$, $5 \times m$, $6 \times n$ arrays. Similarly

CALL MTMPY (A,B,C,5,3,2,7,5,6,1)

forms the product $C_{11} = A_{11}^T B_{11}$, where A_{11} , B_{11} , C_{11} are 3×5 , 3×2 ,
 5×2 matrices stored in $7 \times l$, $5 \times m$, $6 \times n$ arrays.

DMPY-3

\$IBFTC MC13MO XR3,M94,NODD,LIST	
SUBROUTINE DMPY(A,B,C,NRA,NRB,NCB,NA,NB,NC,IP)	DMPL0000
DIMENSION A(1),B(1),C(1)	DMPL0001
DOUBLE PRECISION A,B,C	DMPL0002
1 IF(IP-1)2,2,3	DMPL0003
2 ND=1	DMPL0004
ND1=NB	DMPL0005
IF(IP)4,4,5	DMPL0006
3 ND=NB	DMPL0007
ND1=1	DMPL0008
IF(IP-2)4,4,5	DMPL0009
4 MD=NA	DMPL0010
MD1=1	DMPL0011
GO TO 6	DMPL0012
5 MD=1	DMPL0013
MD1=NA	DMPL0014
6 N1=1	DMPL0015
L1=1	DMPL0016
DO 9 J=1,NCB	DMPL0017
L=L1	DMPL0018
M1=1	DMPL0019
DO 8 I=1,NRA	DMPL0020
C(L)=0.0	DMPL0021
M=M1	DMPL0022
N=N1	DMPL0023
DO 7 K=1,NRB	DMPL0024
C(L)=C(L)+A(M)*B(N)	DMPL0025
M=M+MD	DMPL0026
7 N=N+ND	DMPL0027
M1=M1+MD1	DMPL0028
8 L=L+1	DMPL0029
N1=N1+ND1	DMPL0030
9 L1=L1+NC	DMPL0031
9999 RETURN	DMPL0032
END	

Subroutine: DMVTRN

Purpose: To compute the matrix product of a 3 x 3 matrix and a 3 x N matrix in double precision.

Calling Sequence: CALL DMVTRN (A, B, C, M, N)

Input and Output

I/∅	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	A	d(3, 3)			Pre-matrix.
I	B	d(3, N)			Post-matrix.
∅	C	d(3, N)			Matrix product, AB or A ^T B.
I	M				=1: Pre-matrix is A. =2: Pre-matrix is A ^T .
I	N				Number of columns of B, C.

Common storages used: None

Subroutines required: None

DMVTRN-1

```
$IBFTC MC132J XR3,M94,NODD,LIST
SUBROUTINE DMVTRN (A,B,C,M,N)
DOUBLE PRECISION A(9),B(9),C(9)
INTEGER NN(4)
DATA NN/3,1,8,0/
```

C

```
1 I1 = 1
  J1 = 1
  K1 = 1
  DO 4 I=1,N
    DO 3 J=1,3
      C(I1) = 0.DO
      DO 2 K=1,3
        C(I1) = C(I1)+A(J1)*B(K1)
        J1 = J1+NN(M)
      2 K1 = K1+1
        I1 = I1+1
        J1 = J1-NN(M+2)
    3 K1 = K1-3
      K1 = K1+3
  4 J1 = 1
  RETURN
END
```

```
DMVT0001
DMVT0002
DMVT0003
DMVT0004
DMVT0005
DMVT0006
DMVT0007
DMVT0008
DMVT0009
DMVT0010
DMVT0011
DMVT0012
DMVT0013
DMVT0014
DMVT0015
DMVT0016
DMVT0017
DMVT0018
DMVT0019
DMVT0020
DMVT0021
```

Subroutine: DNØRM

Purpose: Compute the magnitude of a vector.

Calling Sequence: Z = DNØRM (X)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	X	d(3)			Vector
Ø	DNØRM	d			Magnitude of X.

Common storages used: None

Subroutines required: DDØT

DNØRM-1

```
$IBFTC MC132F XR3,M94,NODD,LIST
DOUBLE PRECISION FUNCTION DNORM (X)
DOUBLE PRECISION DDOT,DSQRT
1 DNORM = DSQRT(DDOT(X,X))
RETURN
END
```

```
DNRM0001
DNRM0002
DNRM0003
DNRM0004
```

Subroutine: DØT

Purpose: Computes the inner (dot) product of two 3-vectors.

Calling Sequence: Z = DØT (X,Y)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	X, Y	(3)			Input vectors
Ø	Z				$X \cdot Y$

Common storages used: None

Subroutines required: None

DØT-1

```
SIBFTC DOT      XR3,M94,NODD
  FUNCTION DOT  (X,Y)
  DIMENSION X(3) ,Y(3)
  SUM = 0.
  DO 1 I=1,3
1  SUM = SUM+X(I)*Y(I)
  DOT = SUM
  RETURN
  END
```

```
DOT 0001
DOT 0002
DOT 0003
DOT 0004
DOT 0005
DOT 0006
DOT 0007
DOT 0008
```

Subroutine: DPFMRS

Purpose: To compute planetary positions and velocities.

Calling Sequence: CALL DPFMRS (ET,UCB,ICB,NERR,T)

Input and Output

I/∅	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	ET	d	t_E	sec	Ephemeris time, seconds from 1950 January 0.0.
I	UCB	d(14)			Gravitation constants and ephemeris scale factors.
I	ICB				Central body number
∅	NERR				Error flag.
I∅	T	(4)			Ephemeris tape position indicators.

Common storages used: //22 cells, /DFMCOM/

Subroutines required: None

DPFMRS-1

Reference: "User's Description of JPL Ephemeris Tapes," JPL-TR32-580,
P.R. Peabody, J.F. Scott, E.G. Orozco, March 2, 1964.

1. /DFMCOM/CFM(2232) (See paragraph 3, below.)

Location	Symbol	Dimension	Description
CFM(1)	IFM	14	Output options (input) IFM(i) = 0 ignore i th body 1 output position only 2 output position and velocity
(15)	RFM	d(6,12)	RFM(1-3,i) = i th body position wrt central body RFM(4-6,i) = i th body velocity wrt central body
(159)	DFM	d(4)	DFM(1-2) = nutations in longitude and obliquity DFM(3-4) = nutation rates
(167)	BFM	d(829)	Position-velocity buffer
(1825)	SNT	2,204	Nutation buffer

2. Usage:

The planets are numbered outward from the sun. That is

ICB = 1	Mercury	6	Saturn
2	Venus	7	Uranus
3	Earth	8	Neptune
4	Mars	9	Pluto
5	Jupiter		

In addition,

ICB = 10	Sun
11	Moon
12	Earth-Moon Barycenter

DPFMRS-2

Output of the nutations is keyed by IFM(13). The array IFM must be loaded prior to the first call of DPFMRS. If either the Earth or Moon are keyed for output, it is necessary that both IFM(11),IFM(12) be not less than the maximum of IFM(3),IFM(11),IFM(12). In addition, IFM(ICB) must be not less than the maximum of IFM(i), $i = 1, 2, \dots, 12$.

The array UCB contains, in components 1-11, the gravitational constants of the bodies. UCB(12) is computed from

$$\mu_{12} = \mu_3 + \mu_{11}$$

The heliocentric positions and velocities of the planets, except Earth, and the Barycenter are computed in a.u. and a.u./day, and the geocentric position and velocity of the Moon are computed in earth radii and earth radii/day. For output, they are scaled into km and km/sec using

$$UCB(13) = \text{km/a.u.}$$

$$UCB(14) = \text{km/earth radius}$$

The interpolation formulas are described in the reference.

The array T is used for maintaining knowledge of the ephemeris tape position. Overlay of T will cause unnecessary rewinding of the ephemeris tape. For the initial call of DPFMRS, T(3) must be not greater than 0.

3. Reduced Ephemeris

The dimensions in paragraph 1 are used in decks MC1322(DPFMRS) and MC1324(DFMCOM) is also supplied. The reduced ephemeris is interchangeable with the full ephemeris with the following exceptions:

1. Positions and velocities of Mercury, Uranus, Neptune and Pluto are not computed. The 1, 7, 8, 9 components of IFM are ignored.

DPFMRS-3

2. IFM(13) must not be 2; i.e., no nutation rates may be requested.

3. The dimensions of CFM are

CFM (1524)

BFM d(577)

SNT (2,102)

DPFMRS-4

```

$IBFTC MC1322 XR3,M94,NODD,LIST
SUBROUTINE DPFMRS (ET,UCB,ICB,NERR,T)
PLANETARY EPHEMERIS
USES JPL EPHEMERIS TAPE SYSTEM

DOUBLE PRECISION ET,UCB(14)
DIMENSION T(4)

DOUBLE PRECISION BFM,DFM,RFM
DIMENSION BUFEFM(1658)
COMMON /DFMCOM/IFM(14),RFM(6,12)
1 DFM(4),BFM(829),SNT(2,204)
EQUIVALENCE (BFM(1),BUFEFM,TBUF)

DOUBLE PRECISION D(11)
COMMON S(22)
EQUIVALENCE (S(1),D)

DIMENSION DEL(3),IDF(3),IDL(13,2)
DATA DEL/ 2., 4., .5
1 /,IDF/ 45, 27,153
2 /,IDL/ 1, 2, 2, 2, 2, 2, 2, 2, 0, 3, 0, 3
3 , 2, 92,146,200,254,308,362,416,470, 0,524, 0,830
4 /

INITIALIZATION AND ERROR CHECK
1 D(1) = ET/86400.D0+2433282.5D0
JCB = ICB
IF (T(3).GT.0.) GO TO 4
3 REWIND 8
READ (8) (BUFEFM(I),I=1,24)
READ (8) T
T(1) = T(3)-8.
T(2) = T(3)
4 IERR = 2
IF (S(1).GT.T(4)) GO TO 999
IERR = 1
IF (S(1).LT.T(3)) GO TO 999
IERR = 0

CHECK BUFFER AND READ IN CORRECT 8-DAY RECORD
10 S(3) = S(1)-T(1)
IF (S(3).GE.0.) GO TO 12
IF (S(1).LT.T(2)) GO TO 3
IKK = DEL(1)-S(3)/8.
DO 11 I=1,IKK
11 BACKSPACE 8
GO TO 15
12 IF (S(3).LT.8.) GO TO 14
IKK = S(3)/8.-1.
IF (IKK.EQ.0) GO TO 15
DO 13 I=1,IKK
13 READ (8) TBUF
GO TO 15
14 IF (TBUF.EQ.T(1)) GO TO 16
BACKSPACE 8
15 READ (8) BUFEFM,(SNT(1,I),I=1,204)
T(1) = TBUF
T(2) = TBUF+(T(3)-TBUF)/4.
16 CONTINUE
ILL = 1
17 IKK = IFM(3)
IFM(3) = IFM(12)
IFM(12) = IKK
DO 18 I=1,6
D(2) = RFM(I,3)
RFM(I,3) = RFM(I,12)
18 RFM(I,12) = D(2)
GO TO (20,45),ILL

INTERPOLATION CONTROL
20 D(2) = D(1)-BFM(1)
ILL = 4
NNN = 3
D(10) = UCB(13)

```

```

FMR S0001
FMR S0002
FMR S0003
FMR S0004
FMR S0005
FMR S0006
FMR S0007
FMR S0008
FMR S0009
FMR S0010
FMR S0011
FMR S0012
FMR S0013
FMR S0014
FMR S0015
FMR S0016
FMR S0017
FMR S0018
FMR S0019
FMR S0020
FMR S0021
FMR S0022
FMR S0023
FMR S0024
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FMR S0027
FMR S0028
FMR S0029
FMR S0030
FMR S0031
FMR S0032
FMR S0033
FMR S0034
FMR S0035
FMR S0036
FMR S0037
FMR S0038
FMR S0039
FMR S0040
FMR S0041
FMR S0042
FMR S0043
FMR S0044
FMR S0045
FMR S0046
FMR S0047
FMR S0048
FMR S0049
FMR S0050
FMR S0051
FMR S0052
FMR S0053
FMR S0054
FMR S0055
FMR S0056
FMR S0057
FMR S0058
FMR S0059
FMR S0060
FMR S0061
FMR S0062
FMR S0063
FMR S0064
FMR S0065
FMR S0066
FMR S0067
FMR S0068
FMR S0069
FMR S0070
FMR S0071
FMR S0072
FMR S0073
FMR S0074

```

```

D(11) = D(10)/86400.D0
DO 39 I=1,13
IKK = IFM(I)
IF (IKK.EQ.0) GO TO 39
IF (I.EQ.13) GO TO 22
IF (I.EQ.11) GO TO 21
IF (ILL.EQ.IDL(I,1)) GO TO 30
GO TO 25
21 D(10) = UCB(14)
D(11) = D(10)/86400.D0
GO TO 25
22 D(10) = 1.D0
NNN = 2
IF (ILL.EQ.3) GO TO 30
C
C COMPUTE INTERPOLATION COEFFICIENTS
25 ILL = IDL(I,1)
IF (ILL.EQ.0) GO TO 39
ILC = S(3)/DEL(ILL)
D(6) = D(2)-DEL(ILL)*FLOAT(ILC)
D(3) = 1.D0-D(6)
DO 26 J=3,6,3
D(8) = D(J)*D(J)-1.D0
D(J+1) = D(J)*D(8)/6.D0
26 D(J+2) = D(J+1)*(D(8)-3.D0)/20.D0
C
C INTERPOLATE
30 LOC = IDL(I,2)+3*NNN*ILC
IMM = LOC
INN = IMM+3*NNN
DO 32 K=1,NNN
D(9) = 0.D0
DO 31 J=3,5
D(9) = D(9)+D(J)*BFM(IMM)+D(J+3)*BFM(INN)
IMM = IMM+1
31 INN = INN+1
32 RFM(K,I) = D(10)*D(9)
IF (IKK.EQ.1) GO TO 39
35 IMM = LOC+IDF(ILL)
IF (I.EQ.13) IMM=IMM-51
INN = IMM+3*NNN
DO 38 K=1,NNN
D(9) = 0.D0
DO 36 J=3,5
D(9) = D(9)+D(J)*BFM(IMM)+D(J+3)*BFM(INN)
IMM = IMM+1
36 INN = INN+1
IF (I.EQ.13) GO TO 37
RFM(K+3,I) = D(11)*D(9)
GO TO 38
37 RFM(K+2,I) = D(9)
38 CONTINUE
39 CONTINUE
C
C EARTH/EARTH-MOON BARYCENTER
40 UCB(12) = UCB(3)+UCB(11)
IF (IFM(11).EQ.0) GO TO 42
D(3) = UCB(3)/UCB(12)
D(4) = UCB(11)/UCB(12)
ILL = 3*IFM(11)
DO 41 I=1,ILL
RFM(I,12) = RFM(I,3)-D(4)*RFM(I,11)
41 RFM(I,11) = RFM(I,3)+D(3)*RFM(I,11)
42 ILL = 2
GO TO 17
C
C SUN
45 IF (IFM(10).EQ.0) GO TO 50
ILL = 3*IFM(10)
DO 46 I=1,ILL
46 RFM(I,10) = 0.D0
C
C SHIFT BODY CENTERS
50 IF (JCB.EQ.10) GO TO 999
DO 51 I=1,6

```

```

FMR S0075
FMR S0076
FMR S0077
FMR S0078
FMR S0079
FMR S0080
FMR S0081
FMR S0082
FMR S0083
FMR S0084
FMR S0085
FMR S0086
FMR S0087
FMR S0088
FMR S0089
FMR S0090
FMR S0091
FMR S0092
FMR S0093
FMR S0094
FMR S0095
FMR S0096
FMR S0097
FMR S0098
FMR S0099
FMR S0100
FMR S0101
FMR S0102
FMR S0103
FMR S0104
FMR S0105
FMR S0106
FMR S0107
FMR S0108
FMR S0109
FMR S0110
FMR S0111
FMR S0112
FMR S0113
FMR S0114
FMR S0115
FMR S0116
FMR S0117
FMR S0118
FMR S0119
FMR S0120
FMR S0121
FMR S0122
FMR S0123
FMR S0124
FMR S0125
FMR S0126
FMR S0127
FMR S0128
FMR S0129
FMR S0130
FMR S0131
FMR S0132
FMR S0133
FMR S0134
FMR S0135
FMR S0136
FMR S0137
FMR S0138
FMR S0139
FMR S0140
FMR S0141
FMR S0142
FMR S0143
FMR S0144
FMR S0145
FMR S0146
FMR S0147
FMR S0148
FMR S0149

```

```
51 D(I) = RFM(I,JCB)
DO 53 I=1,12
IF (IFM(I).EQ.0) GO TO 53
ILL = 3*IFM(I)
DO 52 J=1,ILL
52 RFM(J,I) = RFM(J,I)-D(J)
53 CONTINUE
```

```
C
999 NERR = IERR
RETURN
END
```

```
FMR50150
FMR50151
FMR50152
FMR50153
FMR50154
FMR50155
FMR50156
FMR50157
FMR50158
FMR50159
```



```

NNN = 3
D(10) = UCB(13)
D(11) = D(10)/86400.D0
DO 39 I=1,13
IKK = IFM(I)
IF (IKK.EQ.0) GO TO 39
IF (I.EQ.13) GO TO 22
IF (I.EQ.11) GO TO 21
IF (ILL.EQ.IDL(I,1)) GO TO 30
GO TO 25
21 D(10) = UCB(14)
D(11) = D(10)/86400.D0
GO TO 25
22 D(10) = 1.D0
NNN = 2
IF (ILL.EQ.3) GO TO 30
C
C COMPUTE INTERPOLATION COEFFICIENTS
25 ILL = IDL(I,1)
IF (ILL.EQ.0) GO TO 39
ILC = S(3)/DEL(ILL)
D(6) = D(2)-DEL(ILL)*FLOAT(ILC)
D(3) = 1.D0-D(6)
DO 26 J=3,6,3
D(8) = D(J)*D(J)-1.D0
D(J+1) = D(J)*D(8)/6.D0
26 D(J+2) = D(J+1)*(D(8)-3.D0)/20.D0
C
C INTERPOLATE
30 LOC = IDL(I,2)+3*NNN*ILC
IMM = LOC
INN = IMM+3*NNN
DO 32 K=1,NNN
D(9) = 0.D0
DO 31 J=3,5
D(9) = D(9)+D(J)*BFM(IMM)+D(J+3)*BFM(INN)
IMM = IMM+1
31 INN = INN+1
32 RFM(K,I) = D(10)*D(9)
IF (IKK.EQ.1) GO TO 39
35 IMM = LOC+IDF(ILL)
INN = IMM+3*NNN
DO 38 K=1,NNN
D(9) = 0.D0
DO 36 J=3,5
D(9) = D(9)+D(J)*BFM(IMM)+D(J+3)*BFM(INN)
IMM = IMM+1
36 INN = INN+1
RFM(K+3,I) = D(11)*D(9)
38 CONTINUE
39 CONTINUE
C
C EARTH/EARTH-MOON BARYCENTER
40 UCB(12) = UCB(3)+UCB(11)
IF (IFM(11).EQ.0) GO TO 42
D(3) = UCB(3)/UCB(12)
D(4) = UCB(11)/UCB(12)
ILL = 3*IFM(11)
DO 41 I=1,ILL
RFM(I,12) = RFM(I,3)-D(4)*RFM(I,11)
41 RFM(I,11) = RFM(I,3)+D(3)*RFM(I,11)
42 ILL = 2
GO TO 17
C
C SUN
45 IF (IFM(10).EQ.0) GO TO 50
ILL = 3*IFM(10)
DO 46 I=1,ILL
46 RFM(I,10) = 0.D0
C
C SHIFT BODY CENTERS
50 IF (JCB.EQ.10) GO TO 999
DO 51 J=1,6
51 D(J) = RFM(J,JCB)
DO 53 I=1,12
IF (IFM(I).EQ.0) GO TO 53
FMR50075
FMR50076
FMR50077
FMR50078
FMR50079
FMR50080
FMR50081
FMR50082
FMR50083
FMR50084
FMR50085
FMR50086
FMR50087
FMR50088
FMR50089
FMR50090
FMR50091
FMR50092
FMR50093
FMR50094
FMR50095
FMR50096
FMR50097
FMR50098
FMR50099
FMR50100
FMR50101
FMR50102
FMR50103
FMR50104
FMR50105
FMR50106
FMR50107
FMR50108
FMR50109
FMR50110
FMR50111
FMR50112
FMR50113
FMR50114
FMR50115
FMR50116
FMR50117
FMR50118
FMR50119
FMR50120
FMR50121
FMR50122
FMR50123
FMR50124
FMR50125
FMR50126
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FMR50128
FMR50129
FMR50130
FMR50131
FMR50132
FMR50133
FMR50134
FMR50135
FMR50136
FMR50137
FMR50138
FMR50139
FMR50140
FMR50141
FMR50142
FMR50143
FMR50144
FMR50145
FMR50146
FMR50147
FMR50148
FMR50149

```

```
ILL = 3*IFM(I)
DO 52 J=1,ILL
52 RFM(J,I) = RFM(J,I)-D(J)
53 CONTINUE
C
999 NERR = IERR
RETURN
END
```

```
FMRS0150
FMRS0151
FMRS0152
FMRS0153
FMRS0154
FMRS0155
FMRS0156
```

Subroutine: DRAGD

Purpose: To compute the acceleration due to atmospheric drag.

Calling Sequence: CALL DRAGD (R,V,TR,WE,RE,C,A)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	R	d(3)	R	km	Vehicle position relative to central body.
I	V	d(3)	V	km/sec	Vehicle velocity relative to central body.
I	TR	d(3,3)	T		Transformation from EE of date or body-fixed system to base coordinate system.
I	WE	d	ω_E	rad/sec	Central body rotation rate.
I	RE	d	r_E	km	Central body's radius.
I	C	d(2)	C_1, C_2		Drag coefficients.
IØ	A	d(3)	A	km/sec ²	Acceleration.

Common storages used: 12 cells

Subroutines required: DDØT, DCRØSS

DRAGD-1

```

$IBFTC MC132T XR3,M94,NODD,LIST
SUBROUTINE DRAGD (R,V,TR,WE,RF,C,A)
C   ATMOSPHERIC DRAG ACCELERATION
DOUBLE PRECISION A(3),C(2),R(3),TR(3,3),V(3),WE,RF
1   ,D,DDOT,DEXP,DSQRT
COMMON D(6)
C
1 CALL DCROSS (TR(1,3),R,D)
DO 2 I=1,3
2 D(I) = V(I)-WE*D(I)
D(4) = DDOT(R,R)
D(4) = DSQRT(D(4))
D(5) = RE-D(4)
D(6) = DDOT(D,D)
D(6) = DSQRT(D(6))
D(5) = C(2)*D(5)
D(5) =-C(1)*D(6)*DEXP(D(5))
DO 3 I=1,3
3 A(I) = A(I)+D(5)*D(I)
999 RETURN
END
DRAG0001
DRAG0002
DRAG0003
DRAG0004
DRAG0005
DRAG0006
DRAG0007
DRAG0008
DRAG0009
DRAG0010
DRAG0011
DRAG0012
DRAG0013
DRAG0014
DRAG0015
DRAG0016
DRAG0017
DRAG0018
DRAG0019

```

Subroutine: DRAGDP

Purpose: To compute the acceleration due to atmospheric drag, its gradient and partial derivatives with respect to the drag coefficients.

Calling Sequence: CALL DRAGDP (R,V,TR,WE,RE,C,A,AP,G,N)

Input and Output

I/∅	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	R	d(3)	R	km	Vehicle position relative to the central body.
I	V	d(3)	V	km/sec	Vehicle velocity relative to the central body.
I	TR	d(3,3)	T		Transformation from EE of date or body-fixed system to base coordinate system.
I	WE	d	ω_E	rad/sec	Central body rotation rate.
I	RE	d	r_E	km	Central body radius.
I	C	d(2)	C_1, C_2		Drag coefficients.
I∅	A	d(3)	A	km/sec ²	Acceleration.
∅	AP	d(6)			$\partial A / \partial C_1 \quad \partial A / \partial C_2$
I∅	G	d(3,6)	∇A		Gradients $\nabla_R A, \nabla_V A$
I	N				Option switch =0: acceleration only. ≠0: $\partial A / \partial C_i$ and gradient.

Common storages used: 54 cells

Subroutines required: DDOT, DCROSS, DMVTRN

DRAGDP-1

Method:

The expression assumed for atmospheric drag is

$$A = - c_1 e^{-c_2 h} |V_a| V_a$$

where c_1 , c_2 are constants and

h = altitude (= $|R| - r_E$)

V_a = velocity relative to the atmosphere.

Implied assumptions include an exponential atmosphere rotating rigidly with the central body. The first coefficient, c_1 , is

$$c_1 = \frac{1}{2} \rho_0 c_D S/M$$

where

ρ_0 = "sea level" atmospheric density (Earth $\approx 1.225 \text{ kg/m}^3$)

c_D = drag coefficient ($\approx .15 - .20$ for hypersonic velocities)

S = cross-sectional area of vehicle

M = vehicle mass

The second coefficient, c_2 , is

$$c_2 \approx \frac{g_E M_0}{R^* T_M}$$

where

g_E = gravity at central body surface

M_0 = mean molecular weight of atmosphere

DRAGDP-2

R^* = universal gas constant

T_M = molecular temperature of the atmosphere

For Earth, at altitudes above 100 km,

$$C_2 \approx .14$$

The velocity of the vehicle relative to the atmosphere, V_a , is

$$V_a = V - \omega_E \hat{k} \times R$$

where \hat{k} is a unit vector along the central body's north polar axis.

The partial derivatives of A are

$$\frac{\partial A}{\partial C_1} = \frac{A}{C_1}, \quad \text{computed if } N = -1, -3$$

$$\frac{\partial A}{\partial C_2} = -hA, \quad \text{computed if } N = -2, -3$$

The gradients of A are

$$\nabla_R A = -\frac{C_2}{r} AR^T - \omega_E \nabla_V A (\hat{k} \times)$$

$$\nabla_V A = \frac{1}{V_a^2} AV_a^T - C_1 e^{-c_2 h} |V_a| I$$

Both A and its gradients are summed with input values.

DRAGDP-3

SIBFTC MC132U XR3,M94,NODD,LIST	
SUBROUTINE DRAGDP (R,V,TR,WE,RE,C,A,AP,G,N)	
C ATMOSPHERIC DRAG ACCELERATION	
DOUBLE PRECISION A(3),AP(3,2),C(2),G(3,6),R(3),TR(3,3),V(3)	DRGP0001
1 COMMON ,WE,RF,U(5),D,DDOT,DEXP,DSQRT	DRGP0002
DATA D(3,9)	DRGP0003
DATA U/ 0.DO, 0.DO,-1.DO, 0.DO, 0.DO/	DRGP0004
C	DRGP0005
1 DO 2 I=1,3	DRGP0006
2 D(I,2) = WE*TR(I,3)	DRGP0007
CALL DCROSS (D(1,2),R,D(1,1))	DRGP0008
DO 3 I=1,3	DRGP0009
3 D(I,1) = V(I)-D(I,1)	DRGP0010
D(1,7) = DDOT(R,R)	DRGP0011
D(1,7) = DSQRT(D(1,7))	DRGP0012
D(2,7) = RE-D(1,7)	DRGP0013
D(3,7) = DDOT(D,D)	DRGP0014
D(1,8) = DSQRT(D(3,7))	DRGP0015
D(2,8) = C(2)*D(2,7)	DRGP0016
D(2,8) = -D(1,8)*DEXP(D(2,8))	DRGP0017
4 ASSIGN 10 TO LL	DRGP0018
IF (N.GE.0) GO TO 7	DRGP0019
ASSIGN 8 TO LL	DRGP0020
D(1,9) = 1.DO	DRGP0021
D(2,9) = C(1)*D(2,7)	DRGP0022
IF (N+2) 7,5,6	DRGP0023
5 D(1,9) = D(2,9)	DRGP0024
6 ASSIGN 9 TO LL	DRGP0025
7 CONTINUE	DRGP0026
DO 10 I=1,3	DRGP0027
D(I,3) = D(2,8)*D(I,1)	DRGP0028
A(I) = A(I)+C(1)*D(I,3)	DRGP0029
GO TO LL, (8,9,10)	DRGP0030
8 AP(I,2) = D(2,9)*D(I,3)	DRGP0031
9 AP(I,1) = D(1,9)*D(I,3)	DRGP0032
10 CONTINUE	DRGP0033
IF (N.EQ.0) GO TO 999	DRGP0034
11 D(2,7) = C(1)*D(2,8)	DRGP0035
D(1,7) = -C(1)*C(2)/D(1,7)	DRGP0036
DO 13 I=1,3	DRGP0037
D(1,8) = D(I,3)*D(1,7)	DRGP0038
D(2,8) = D(I,3)/D(3,7)	DRGP0039
DO 12 J=1,3	DRGP0040
D(I,J+3) = D(2,8)*D(J,1)	DRGP0041
12 G(I,J) = G(I,1)+D(1,8)*R(J)	DRGP0042
13 D(I,I+3) = D(I,I+3)+D(2,7)	DRGP0043
J = 9	DRGP0044
DO 14 I=1,3	DRGP0045
CALL DCROSS (D(1,2),U(I),D(1,J))	DRGP0046
14 J = J-1	DRGP0047
CALL DMVTRN (D(1,4),D(1,7),D(1,1),1,3)	DRGP0048
DO 15 I=1,3	DRGP0049
DO 15 J=1,6	DRGP0050
15 G(I,J) = G(I,J)+D(I,J)	DRGP0051
999 RETURN	DRGP0052
END	DRGP0053
	DRGP0054

Subroutine: DSDAT

Purpose: Computes three DSIF (JPL) measurements (hour angle, declination, doppler).

Calling Sequence: CALL DSDAT(GHA)

Input and Output

I/ ϕ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	GHA	d	φ	rad	Greenwich hour angle at time of measurements.

Common storages used: /DATCOM/

Subroutines required: DCR0SS, DD0T, DMVTRN, DVN0RM

DSDAT-1

COMMON LOCATIONS

Common	Location	Name	Dimension	Description
DATCOM	C(1)	BIAS	d(2)	BIAS(1), Doppler bias frequency (cps) BIAS(2), Transponder retransmission ratio.
	C(133)	FTR	d	Doppler transmitter frequency (cps).
	C(135)	ØMEGA	d	Earth rotation rate (rad/sec).
	C(137)	SPDLT	d	Speed of light (km/sec).
	C(139)	STA	d(10)	STA(1-3), Receiving station position in B-frame (km). STA (4-5) Refraction constants. STA(6-10), Same as above, but for transmitting station if doppler is three-way.
	C(159)	TAU	d	Doppler count interval (sec).
	C(161)	TB2CØ	d(18)	TB2CØ(1-9), B-frame to C-frame transform at end of doppler count interval at signal reception. TB2CØ(10-18), As above, but at beginning of doppler count interval.
	C(197)	TB2CT	d(18)	Same as TB2CØ, but at time of signal transmission.
	C(233)	TT2BØ	d(9)	Unit North, East, Down vectors in B-frame at receiving station.
	C(251)	TT2BT	d(9)	Same as TT2BØ, but for transmitting when in three-way doppler mode.
	C(269)	XV	d(12)	XV(1-6), Spacecraft position and velocity at end of doppler interval as signal leaves spacecraft. XV(7-12), Same as above, but at beginning of count interval.
	C(294)	MØDE		Doppler mode key +2, two way +3, three way
	C(298)	NANG		Angle inclusion key*.
	C(299)	NFRAC		Refraction correction key*. *+1, include. +2, omit.
	C(5)	ØBS	d(64)	Measurements. See Table 1, description of CBDAT.

DSDAT-2

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SIBFTC MC13DM M94,NODD,XR3
CMC13DM DSIF MEASUREMENTS
SUBROUTINE DSDAT(GHA)
COMMON /DATCOM/ BIAS(2), OBS(64), FTR, OMEGA
1, SPDLT, STA(10), TAU, TB2CO(18)
2, TB2CT(18), TT2BO(9), TT2BT(9), XV(12)
3, MLT, MODE, MSTA, MTIM
4, NALIGN, NANG, NFRAC
DOUBLE PRECISION DCXSOB(3), DCXSTR(3), DELTE(2), DNOB(3)
1, DNTR(3), DRDT(2), DUM1(3), DUM2(3)
2, E(2), EDOT(2), EK(3), OMG(3)
3, RM(2), RTMG(2), RTOB(6), RTOBX(6)
4, RTTR(6), RTTRX(6), SCEK(3), SXOB(3)
5, SXTR(9), TT2CO(9), TT2CT(9), XSOB(6)
6, XSOBX(6), XSTR(6), XSTRX(6), Z1(3)
7, Z2(3)
DOUBLE PRECISION BIAS, CF, CONST, CX, CY
1, OBS, DDOT, DEN1, DEN2
2, DNORM, F, FTR, GHA, OMEGA
3, RAOB, RAOBX, RATTR, RATTRX, RSPLT
4, RSPLT2, R2OB, R2TR, SA, SB
5, SE, SLON, SPDLT, STA, S1
6, S2, T, TAU, TB2CO, TB2CT
7, TT2BO, TT2BT, XTDOB, XTDT, XTEOB
8, XTETR, XTNOB, XTNTR, XV
DOUBLE PRECISION AZ, CPHI, SPHI
C
C DECLARE LIBRARY FUNCTIONS DOUBLE PRECISION TO SATISFY UNIVAC
DOUBLE PRECISION DATAN2,DATAN,DCOS,DSIN,DSQRT
C
EQUIVLFNCE (SXOR(1),XTNOB), (SXOB(2),XTEOB), (SXOB(3),XTDOB)
1, (SXTR(1),XTNTR), (SXTR(2),XTETR), (SXTR(3),XTDT)
2, (TT2CO(7),DNOB), (TT2CT(7),DNTR), (RM(1),RAOB)
3, (RM(2),RATTR)
IF(MODE.EQ.3) GO TO 30
DO 10 I=1,5
10 STA(I+5)=STA(I)
DO 20 I=1,9
20 TT2BT(I)=TT2BO(I)
30 RSPLT=1.D0/SPDLT
RSPLT2=RSPLT*RSPLT
C N,E,D VECTORS IN C FRAME AT RECEIVING AND TRANSMITTING TIMES
CALL DMVTRN(TB2CO,TT2BO,TT2CO,1,3)
CALL DMVTRN(TB2CT,TT2BT,TT2CT,1,3)
C STATION VECTORS AT START AND END OF DOPPLER COUNT INTERVAL
CALL DMVTRN(TB2CO,STA,RTOB,1,1)
CALL DMVTRN(TB2CO(10),STA,RTORX,1,1)
CALL DMVTRN(TB2CT,STA(6),RTTR,1,1)
CALL DMVTRN(TB2CT(10),STA(6),RTTRX,1,1)
DO 40 I=1,3
40 OMG(I)=TR2CO(I+6)*OMEGA
CALL DCROSS(OMG,RTOB,RTOB(4))
CALL DCROSS(OMG,RTTR,RTTR(4))
CALL DCROSS(OMG,RTORX,RTOBX(4))
CALL DCROSS(OMG,RTTRX,RTTRX(4))
C RANGE AND RANGE-RATE VECTORS AND MAGNITUDES
DO 50 I=1,6
XSOB(I)=XV(I)-RTOB(I)
XSTR(I)=XV(I)-RTTR(I)
XSOBX(I)=XV(I+6)-RTOBX(I)
50 XSTRX(I)=XV(I+6)-RTTRX(I)
R2OB=DDOT(XSOB,XSOB)
R2TR=DDOT(XSTR,XSTR)
RAOB=DSQRT(R2OB)
RATTR=DSQRT(R2TR)
RAOBX=DNORM(XSOBX)
RATTRX=DNORM(XSTRX)
C RANGE VECTORS IN TOPOCENTRIC COORDINATES
CALL DMVTRN(TT2CO,XSOB,SXOB,2,1)
CALL DMVTRN(TT2CT,XSTR,SXTR,2,1)
C MAGNITUDE OF STATION VECTOR
RTMG(1)=DNORM(RTOB)
RTMG(2)=DNORM(RTTR)
GO TO (60,70),NANG
C HOUR ANGLE AND DECLINATION

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DSDT0001
DSDT0002
DSDT0003
DSDT0004
DSDT0005
DSDT0006
DSDT0007
DSDT0008
DSDT0009
DSDT0010
DSDT0011
DSDT0012
DSDT0013
DSDT0014
DSDT0015
DSDT0016
DSDT0017
DSDT0018
DSDT0019
DSDT0020
DSDT0021
DSDT0022
DSDT0023
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DSDT0028
DSDT0029
DSDT0030
DSDT0031
DSDT0032
DSDT0033
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DSDT0036
DSDT0037
DSDT0038
DSDT0039
DSDT0040
DSDT0041
DSDT0042
DSDT0043
DSDT0044
DSDT0045
DSDT0046
DSDT0047
DSDT0048
DSDT0049
DSDT0050
DSDT0051
DSDT0052
DSDT0053
DSDT0054
DSDT0055
DSDT0056
DSDT0057
DSDT0058
DSDT0059
DSDT0060
DSDT0061
DSDT0062
DSDT0063
DSDT0064
DSDT0065
DSDT0066
DSDT0067
DSDT0068
DSDT0069
DSDT0070
DSDT0071
DSDT0072
DSDT0073
DSDT0074

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60 SLON=DATAN2(STA(2),STA(1))
ORS(1)=GHA+SLON-DATAN2(XSOB(2),XSOB(1))
CALL DVNORM(TB2CO(7),EK)
CALL DCROSS(XSOB,EK,SCEK)
S2=DDOT(SCEK,SCEK)
S1=DSQRT(S2)
OBS(17)=DATAN(XSOR(3)/S1)
C DIFFERENCED DOPPLER OBSFRVABLE
70 CONST=BIAS(2)*FTR*RSPLT
OBS(49)=BIAS(1)*TAU+CONST*(RATOB+RATTR-RATOBX-RATTRX)
GO TO (80,170),NFRAC
C REFRACTION CORRECTIONS
C FIRST, ELEVATIONS AND ELEVATION CORRECTIONS
80 E(1)=DATAN(-XTDOB/DSQRT(XTNOB*XTNOB+XTEOB*XTEOB))
E(2)=DATAN(-XTDTR/DSQRT(XTNTR*XTNTR+XTETR*XTETR))
DO 120 I=1,2
DELTE(I)=0.
IF(E(I).LT..01) GO TO 120
SE=DSIN(E(I))
CE=DCOS(E(I))
IF(E(I).GE..17453793D0) GO TO 110
T=(1.03585796D0-(.01072014D0-(.1279119D-7-.1227363D-7/E(I))/E(I)))/
1E(I)*STA(5*I-1)*CE/SE
90 F=RTMG(I)/RM(I)
100 DELTE(I)=T-F*((STA(5*I-1)+T*T/2.)*CE-T*SE)
GO TO 120
110 DELTE(I)=STA(5*I-1)*CE/SE
120 CONTINUE
CALL DCROSS(DNOB,XSOB,DCXSOB)
GO TO (130,140),NANG
C ANGLE CORRECTIONS
130 SPHI=RTOB(3)/RTMG(1)
CPHI=DSQRT(RTOB(2)**2+RTOB(3)**2)/RTMG(1)
AZ=DATAN2(XTEOB,XTNOB)
OBS(1)=ORS(1)+DSIN(OBS(1))*2*CPHI/(DSIN(AZ)*DCOS(E(1))*2)*DELTE(
11)
OBS(17)=OBS(17)+(DCOS(E(1))*SPHI-DCOS(AZ)*DSIN(E(1))*CPHI)/DCOS(OB
15(17))*DELTE(1)
C DOPPLER CORRECTION
140 CALL DCROSS(DNTR,XSTR,DCXSTR)
CALL DCROSS(DCXSOB,XSOR,DUM1)
CALL DCROSS(DCXSTR,XSTR,DUM2)
DEN1=R2OR*DNORM(DCXSOB)
DEN2=R2TR*DNORM(DCXSTR)
CALL DCROSS(XSOB,OMG,Z1)
CALL DCROSS(XSTR,OMG,Z2)
DO 150 I=1,3
Z1(I)=Z1(I)+XSOB(I+3)
Z2(I)=Z2(I)+XSTR(I+3)
DUM1(I)=DUM1(I)/DEN1
150 DUM2(I)=DUM2(I)/DEN2
EDOT(1)=DDOT(DUM1,Z1)
EDOT(2)=DDOT(DUM2,Z2)
DO 160 I=1,2
SA=DSIN(E(I)+DELTE(I))
SB=DSIN(F(I)+DELTE(I)-EDOT(I)*TAU)
160 DRDT(I)=STA(5*I-1)/STA(5*I)*(1.D-3/SA-1.D-3/SB)
OBS(49)=OBS(49)+CONST*(DRDT(1)+DRDT(2))
170 GO TO (180,190),NANG
180 CONTINUE
IF(OBS(1).LT.0.D0) OBS(1)=OBS(1)+6.2831853071795864
IF(OBS(17).LT.0.D0) OBS(17)=OBS(17)+6.2831853071795864
IF(OBS(1).GT.6.2831853071795864) OBS(1)=OBS(1)-6.2831853071795864
190 RETURN
END
DSDT0075
DSDT0076
DSDT0077
DSDT0078
DSDT0079
DSDT0080
DSDT0081
DSDT0082
DSDT0083
DSDT0084
DSDT0085
DSDT0086
DSDT0087
DSDT0088
DSDT0089
DSDT0090
DSDT0091
DSDT0092
DSDT0093
DSDT0094
DSDT0095
DSDT0096
DSDT0097
DSDT0098
DSDT0099
DSDT0100
DSDT0101
DSDT0102
DSDT0103
DSDT0104
DSDT0105
DSDT0106
DSDT0107
DSDT0108
DSDT0109
DSDT0110
DSDT0111
DSDT0112
DSDT0113
DSDT0114
DSDT0115
DSDT0116
DSDT0117
DSDT0118
DSDT0119
DSDT0120
DSDT0121
DSDT0122
DSDT0123
DSDT0124
DSDT0125
DSDT0126
DSDT0127
DSDT0128
DSDT0129
DSDT0130
DSDT0131
DSDT0132
DSDT0133
DSDT0134
DSDT0135
DSDT0136
DSDT0137
DSDT0138

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Subroutine: DSDATP

Purpose: Computes three DSIF measurements and their partials
(hour angle, declination, doppler).

Calling Sequence: CALL DSDATP(GHA,R2).

Input and Output

I/ ϕ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	GHA	d	φ	rad	Greenwich hour angle at time of measurements.
ϕ	R2	d	S^2	km ²	Slant range squared.

Common storages used: /DATCOM/

Subroutines required: DCR~~O~~SS, DD~~O~~T, DMVTRN, DN~~O~~RM, DVN~~O~~RM

DSDATP- 1

COMMON LOCATIONS

Common	Location	Name	Dimension	Description
DATCOM	C(1)	BIAS	d(2)	BIAS(1), Doppler bias frequency (cps). BIAS(2), Transponder retransmission ratio.
	C(133)	FTR	d	Doppler transmitter frequency (cps).
	C(135)	OMEGA	d	Earth rotation rate (rad/sec).
	C(137)	SPDLT	d	Speed of light (km/sec).
	C(139)	STA	d(10)	STA(1-3), Receiving station position in B-frame (km). STA(4-5), Refraction constants. STA(6-10), As above, but for transmitting station if doppler mode is three-way.
	C(159)	TAU	d	Doppler count interval.
	C(161)	TB2CØ	d(18)	TB2CØ(1-9), B-frame to C-frame transform at end of doppler count interval at signal reception. TB2CØ(10-18), As above, but at beginning of doppler count interval.
	C(197)	TB2CT	d(18)	Same as TB2CØ, but at time of signal transmission.
	C(233)	TT2BØ	d(9)	Unit North, East, Down vectors at receiving station in B-frame.
	C(251)	TT2BT	d(9)	Same as TT2BØ, but for transmitting station if doppler mode is three-way.
	C(269)	XV	d(12)	XV(1-6), Spacecraft position and velocity at end of doppler count interval as signal leaves spacecraft. XV(7-12), as above, but at beginning of doppler count.
	C(293)	MLT		Speed of light partial option key*.
	C(294)	MODE		Doppler mode +2, two-way +3, three-way
	C(295)	MSTA		Station location partial option key*.
	C(296)	MTIM		Station clock partial option key*.

DSDATP-2

COMMON LOCATIONS

Common	Location	Name	Dimension	Description
DATCØM	C(298)	NANG		Angle inclusion option key*.
	C(299)	NFRAC		Refraction correction option key*. *+1, include +2, omit If angles are omitted, no angle partials are computed.
	C(5)	ØBS	d(64)	Measurements See Table 1, description of CBDAT.

DSDATP-3

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$IBFTC MC13DN XR3,M94,NODD,LIST
CMC13DN DSIF MEASUREMENTS AND PARTIALS
SUBROUTINE DSDATP(GHA,R2OB)
COMMON /DATCOM/ BIAS(2), OBS(64), FTR, OMEGA
1, SPDLT, STA(10), TAU, TB2CO(18)
2, TB2CT(18), TT2BO(9), TT2BT(9), XV(12)
3, MLT, MODE, MSTA, MTIM
4, NALIGN, NANG, NFRAC
DOUBLE PRECISION DCXSOR(3), DCXSTR(3), DELTE(2), DNOB(3)
1, DNTR(3), DRDT(2), DUM1(3), DUM2(3)
2, E(2), EDOT(2), EK(3), OMG(3)
3, RM(2), RTMG(2), RTOB(6), RTOBX(6)
4, RTTR(6), RTTRX(6), SCEK(3), SXOB(3)
5, SXTR(3), TT2CO(9), TT2CT(9), XSOB(6)
6, XSOBX(6), XSTR(6), XSTRX(6), Z1(3)
7, Z2(3)
DOUBLE PRECISION BIAS, CF, CONST, CX, CY
1, OBS, DDOT, DEN1, DEN2
2, DNORM, F, FTR, GHA, OMEGA
3, RATOB, RTOBX, RATTR, RATTRX, RSPLT
4, RSPLT2, R2OB, R2TR, SA, SB
5, SE, SLON, SPDLT, STA, S1
6, S2, T, TAU, TB2CO, TB2CT
7, TT2BO, TT2BT, XTDOB, XTDT, XTEOB
8, XTETR, XTNOB, XTNTR, XV
DOUBLE PRECISION AZ, CPHI, SPHI
C
C DECLARE LIBRARY FUNCTIONS DOUBLE PRECISION TO SATISFY UNIVAC
DOUBLE PRECISION DATAN2,DATAN,DCOS,DSIN,DSQRT
C
EQUIVALENCE (SXOB(1),XTNOB), (SXOB(2),XTEOB), (SXOB(3),XTDOB)
1, (SXTR(1),XTNTR), (SXTR(2),XTETR), (SXTR(3),XTDTR)
2, (TT2CO(7),DNOB), (TT2CT(7),DNTR), (RM(1),RATOB)
3, (RM(2),RATTR)
IF(MODE.EQ.3) GO TO 30
DO 10 I=1,5
10 STA(I+5)=STA(I)
DO 20 I=1,9
20 TT2BT(I)=TT2BO(I)
30 RSPLT=1.00/SPDLT
RSPLT2=RSPLT*RSPLT
C
N,E,D VECTORS IN C FRAME AT RECEIVING AND TRANSMITTING TIMES
CALL DMVTRN(TB2CO,TT2BO,TT2CO,1,3)
CALL DMVTRN(TB2CT,TT2BT,TT2CT,1,3)
C
STATION VECTORS AT START AND END OF DOPPLER COUNT INTERVAL
CALL DMVTRN(TB2CO,STA,RTOB,1,1)
CALL DMVTRN(TB2CO(10),STA,RTOBX,1,1)
CALL DMVTRN(TB2CT,STA(6),RTTR,1,1)
CALL DMVTRN(TB2CT(10),STA(6),RTTRX,1,1)
DO 40 I=1,3
40 OMG(I)=TB2CO(I+6)*OMEGA
CALL DCROSS(OMG,RTOB,RTOB(4))
CALL DCROSS(OMG,RTTR,RTTR(4))
CALL DCROSS(OMG,RTOBX,RTOBX(4))
CALL DCROSS(OMG,RTTRX,RTTRX(4))
C
RANGE AND RANGE-RATE VECTORS AND MAGNITUDES
DO 50 I=1,6
XSOB(I)=XV(I)-RTOB(I)
XSTR(I)=XV(I)-RTTR(I)
XSOBX(I)=XV(I+6)-RTOBX(I)
50 XSTRX(I)=XV(I+6)-RTTRX(I)
R2OB=DDOT(XSOB,XSOB)
R2TR=DDOT(XSTR,XSTR)
RATOB=DSQRT(R2OB)
RATTR=DSQRT(R2TR)
RATOBX=DNORM(XSOBX)
RATTRX=DNORM(XSTRX)
C
RANGE VECTORS IN TOPOCENTRIC COORDINATES
CALL DMVTRN(TT2CO,XSOB,SXOB,2,1)
CALL DMVTRN(TT2CT,XSTR,SXTR,2,1)
C
MAGNITUDE OF STATION VECTOR
RTMG(1)=DNORM(RTOB)
RTMG(2)=DNORM(RTTR)
GO TO (60,70),NANG
C
HOUR ANGLE AND DECLINATION

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DSDP0001
DSDP0002
DSDP0003
DSDP0004
DSDP0005
DSDP0006
DSDP0007
DSDP0008
DSDP0009
DSDP0010
DSDP0011
DSDP0012
DSDP0013
DSDP0014
DSDP0015
DSDP0016
DSDP0017
DSDP0018
DSDP0019
DSDP0020
DSDP0021
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DSDP0054
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DSDP0057
DSDP0058
DSDP0059
DSDP0060
DSDP0061
DSDP0062
DSDP0063
DSDP0064
DSDP0065
DSDP0066
DSDP0067
DSDP0068
DSDP0069
DSDP0070
DSDP0071
DSDP0072
DSDP0073
DSDP0074

```

60	SLON=DATAN2(STA(2),STA(1))	DSDP0075
	OBS(1)=GHA+SLON-DATAN2(XSOB(2),XSOB(1))	DSDP0076
	CALL DVNORM(TB2CO(7),EK)	DSDP0077
	CALL DCROSS(XSOB,EK,SCEK)	DSDP0078
	S2=DDOT(SCEK,SCEK)	DSDP0079
	S1=DSQRT(S2)	DSDP0080
	OBS(17)=DATAN(XSOR(3)/S1)	DSDP0081
C	DIFFERENCED DOPPLER OBSERVABLE	DSDP0082
70	CONST=BIAS(2)*FTR*RSPLT	DSDP0083
	OBS(49)=BIAS(1)*TAU+CONST*(RATOB+RATTR-RATOBX-RATTRX)	DSDP0084
	GO TO (80,170),NFRAC	DSDP0085
C	REFRACTION CORRECTIONS	DSDP0086
C	FIRST, ELEVATIONS AND ELEVATION CORRECTIONS	DSDP0087
80	E(1)=DATAN(-XTDOB/DSQRT(XTNOB*XTNOB+XTEOB*XTEOB))	DSDP0088
	E(2)=DATAN(-XTDTR/DSQRT(XTNTR*XTNTR+XTETR*XTETR))	DSDP0089
	DO 120 I=1,2	DSDP0090
	DELTE(I)=0.	DSDP0091
	IF(E(I).LT..01) GO TO 120	DSDP0092
	SE=DSIN(E(I))	DSDP0093
	CE=DCOS(E(I))	DSDP0094
	IF(E(I).GE..17453293D0) GO TO 110	DSDP0095
	T=(1.03585796D0-(.01072014D0-(.1279119D-7-.1227363D-7/E(I))/E(I))/	DSDP0096
	E(I))*STA(5*I-1)*CE/SE	DSDP0097
90	F=RTMG(I)/RM(I)	DSDP0098
100	DELTE(I)=T-F*((STA(5*I-1)+T*T/2.)*CE-T*SE)	DSDP0099
	GO TO 120	DSDP0100
110	DELTE(I)=STA(5*I-1)*CE/SE	DSDP0101
120	CONTINUE	DSDP0102
	CALL DCROSS(DNOR,XSOB,DCXSOB)	DSDP0103
	GO TO (130,140),NANG	DSDP0104
C	ANGLE CORRECTIONS	DSDP0105
130	SPhi=RTOB(3)/RTMG(1)	DSDP0106
	CPhi=DSQRT(RTOB(2)**2+RTOB(3)**2)/RTMG(1)	DSDP0107
	AZ=DATAN2(XTEOB,XTNOB)	DSDP0108
	OBS(1)=OBS(1)+DSIN(OBS(1))*2*CPhi/(DSIN(AZ)*DCOS(E(1))*2)*DELTE(DSDP0109
	11)	DSDP0110
	OBS(17)=OBS(17)+(DCOS(E(1))*SPhi-DCOS(AZ)*DSIN(E(1))*CPhi)/DCOS(OB	DSDP0111
	1S(17))*DELTE(1)	DSDP0112
C	DOPPLER CORRECTION	DSDP0113
140	CALL DCROSS(DNTR,XSTR,DCXSTR)	DSDP0114
	CALL DCROSS(DCXSOB,XSOB,DUM1)	DSDP0115
	CALL DCROSS(DCXSTR,XSTR,DUM2)	DSDP0116
	DEN1=R2OB*DNORM(DCXSOB)	DSDP0117
	DEN2=R2TR*DNORM(DCXSTR)	DSDP0118
	CALL DCROSS(XSOB,OMG,Z1)	DSDP0119
	CALL DCROSS(XSTR,OMG,Z2)	DSDP0120
	DO 150 I=1,3	DSDP0121
	Z1(I)=Z1(I)+XSOR(I+3)	DSDP0122
	Z2(I)=Z2(I)+XSTR(I+3)	DSDP0123
	DUM1(I)=DUM1(I)/DFN1	DSDP0124
150	DUM2(I)=DUM2(I)/DEN2	DSDP0125
	EDOT(1)=DDOT(DUM1,Z1)	DSDP0126
	EDOT(2)=DDOT(DUM2,Z2)	DSDP0127
	DO 160 I=1,2	DSDP0128
	SA=DSIN(E(I)+DELTF(I))	DSDP0129
	SB=DSIN(E(I)+DELTF(I)-EDOT(I)*TAU)	DSDP0130
160	DRDT(I)=STA(5*I-1)/STA(5*I)*(1.0-3/SA-1.0-3/SB)	DSDP0131
	OBS(49)=OBS(49)+CONST*(DRDT(1)+DRDT(2))	DSDP0132
C	PARTIAL DERIVATIVES	DSDP0133
170	GO TO (180,260),NANG	DSDP0134
C	ANGLE PARTIALS	DSDP0135
C	PARTIALS WRT VEHICLE STATE	DSDP0136
180	CONTINUE	DSDP0137
	IF(OBS(1).LT.0.D0) OBS(1)=OBS(1)+6.2831853071795864	DSDP0138
	IF(OBS(17).LT.0.D0) OBS(17)=OBS(17)+6.2831853071795864	DSDP0139
	IF(OBS(1).GT.6.2831853071795864) OBS(1)=OBS(1)-6.2831853071795864	DSDP0140
	CALL DCROSS(SCEK,XSOB,OBS(18))	DSDP0141
	DEN1=R2OB*S1	DSDP0142
	DO 190 I=1,3	DSDP0143
	OBS(I+1)=SCEK(I)/S2	DSDP0144
	OBS(I+4)=0.D0	DSDP0145
	OBS(I+17)=OBS(I+17)/DEN1	DSDP0146
190	OBS(I+20)=0.D0	DSDP0147
C	PARTIALS WRT MEASUREMENT BIAS	DSDP0148
	OBS(8)=1.D0	DSDP0149

```

OBS(24)=1.D0
GO TO (200,210),MTIM
C PARTIALS WRT OBSERVING STATION CLOCK BIAS
200 OBS(9)=DDOT(OBS(2),XSQB(4))+OMEGA
ORS(25)=DDOT(OBS(18),XSQB(4))
210 GO TO (220,230),MLT
C PARTIALS WRT SPEED OF LIGHT
220 OBS(10)=DDOT(OBS(2),XV(4))*RATOB*RSPLT2
OBS(26)=DDOT(OBS(18),XV(4))*RATOB*RSPLT2
230 GO TO (240,260),MSTA
C PARTIALS WRT STATION LOCATION ERRORS
240 CALL DCROSS(EK,RTOB,DUM1)
DEN1=DDOT(DUM1,DUM1)
CALL DMVTRN(TB2CO,OBS(18),OBS(27),2,1)
DO 250 I=1,3
OBS(I+13)=0.D0
OBS(I+29)=0.D0
DUM2(I)=DUM1(I)/DEN1-OBS(I+1)
250 OBS(I+26)=-OBS(I+26)
CALL DMVTRN(TB2CO,DUM2,OBS(11),2,1)
C DOPPLER OBSERVABLE PARTIALS
C PARTIALS WRT VEHICLE STATE
C NORMALIZE RANGE VECTORS
260 DO 270 I=1,3
XSOB(I)=XSOB(I)/RATOB
XSTR(I)=XSTR(I)/RATTR
XSOBX(I)=XSOBX(I)/RATORX
270 XSTRX(I)=XSTRX(I)/RATTRX
C PARTIALS WRT TRANSMITTING STATION LOCATION ARE USED IN THE VEHICLE
C STATE PARTIALS AND WILL BE CALCULATED HERE
CALL DCROSS(OMG,XSTRX,ORS(62))
CALL DCROSS(OMG,ORS(62),DUM1)
CALL DCROSS(OMG,DUM1,DUM2)
DO 280 I=1,3
280 DUM2(I)=(TAU*(ORS(I+61)+.5D0*TAU*(DUM1(I)+TAU*DUM2(I)/3.D0))+XSTRX
1(I)-XSTR(I))*CONST
CX=DDOT(DUM2,RTTR(4))*RSPLT
CY=DDOT(DUM2,RTTRX(4))*RSPLT
DO 290 I=1,3
ORS(I+49)=(XSOB(I)+XSTR(I))*(CONST+CX)-(XSOBX(I)+XSTRX(I))*(CONST+
1CY)
290 ORS(I+52)=CONST*TAU*(XSOBX(I)+XSTRX(I))
C PARTIALS WRT MEASUREMENT BIAS
OBS(56)=TAU
GO TO (300,310),MTIM
C PARTIAL WRT OBSERVING STATION CLOCK BIAS
300 ORS(57)=CONST*(DDOT(XSOB,XSOB(4))+DDOT(XSTR,XSTR(4))-DDOT(XSOBX,X
1ORX(4))-DDOT(XSTRX,XSTRX(4)))
310 GO TO (320,330),MLT
C PARTIAL WRT SPEED OF LIGHT
320 ORS(58)=RSPLT2*(DDOT(ORS(50),XV(4))*RATOB-DDOT(ORS(50),XV(10))*RAT
1ORX+DDOT(DUM2,RTTR(4))*(RATOB+RATTR)-DDOT(DUM2,RTTRX(4))*(RATORX+R
2ATTRX))-CONST*RSPLT*(RATOB+RATTR-RATORX-RATTRX)
330 GO TO (340,370),MSTA
C PARTIALS WRT STATION LOCATION ERRORS
340 CALL DMVTRN(TB2CT,DUM2,ORS(62),2,1)
CALL DCROSS(OMG,XSOBX,ORS(59))
CALL DCROSS(OMG,ORS(59),DUM1)
CALL DCROSS(OMG,DUM1,DUM2)
DO 350 I=1,3
350 DUM2(I)=(TAU*(ORS(I+58)+.5D0*TAU*(DUM1(I)+TAU*DUM2(I)/3.D0))+XSOBX
1(I)-XSOB(I))*CONST
CALL DMVTRN(TB2CO,DUM2,OBS(59),2,1)
IF (MODE.EQ.3) GO TO 370
C ADD STATION LOCATION PARTIALS FOR TWO-WAY DOPPLER
DO 360 I=59,61
360 OBS(I)=ORS(I)+OBS(I+3)
370 RETURN
END
DSDP0150
DSDP0151
DSDP0152
DSDP0153
DSDP0154
DSDP0155
DSDP0156
DSDP0157
DSDP0158
DSDP0159
DSDP0160
DSDP0161
DSDP0162
DSDP0163
DSDP0164
DSDP0165
DSDP0166
DSDP0167
DSDP0168
DSDP0169
DSDP0170
DSDP0171
DSDP0172
DSDP0173
DSDP0174
DSDP0175
DSDP0176
DSDP0177
DSDP0178
DSDP0179
DSDP0180
DSDP0181
DSDP0182
DSDP0183
DSDP0184
DSDP0185
DSDP0186
DSDP0187
DSDP0188
DSDP0189
DSDP0190
DSDP0191
DSDP0192
DSDP0193
DSDP0194
DSDP0195
DSDP0196
DSDP0197
DSDP0198
DSDP0199
DSDP0200
DSDP0201
DSDP0202
DSDP0203
DSDP0204
DSDP0205
DSDP0206
DSDP0207
DSDP0208
DSDP0209
DSDP0210
DSDP0211
DSDP0212
DSDP0213
DSDP0214
DSDP0215
DSDP0216
DSDP0217

```

Subroutine: DSTAT

Purpose: Finds tracking-station position vector in body-fixed coordinates and finds the orthogonal topocentric to body-frame transformation.

Calling Sequence: CALL DSTAT(A,B,DTR,STA,SN)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	A	d	a	km	Body semi-major axis.
I	B	d	b	km	Body semi-minor axis.
I	DTR	d		rad/deg	Degree-to-radian conversion factor.
I	STA	d(3)		deg deg meters	STA(1) = Station latitude (geodetic) STA(2) = Station longitude STA(3) = Station altitude
Ø	SN	d(12)		km, dimension-less	SN(1-3) Station position in body-fixed coordinates. SN(4-12) Topocentric to body-fixed orthogonal transformation.

Common storages used: None

Subroutines required: None

DSTAT-1

Coordinate Frames:

The body-fixed coordinate frame is a right-handed Cartesian (dextral orthogonal) system with the first axis lying in the body equator plane in the direction of the prime meridian, the third axis lying along a line from the body center through the north pole, and the second axis chosen so as to complete the right-handed system.

The topocentric coordinate frame is a dextral orthogonal system. It may be called a local tangent plane system. The first axis lies along the local North vector, the second along the local East vector, and the third along the local Down vector. (The local Down vector at a point on the surface of the body is parallel to the radius vector from the body center to that point if and only if the body is spherical, except for the two poles and all points in the equator.)

Tracker Position Vector:

Let:

- θ be the longitude of the tracker
- ϕ be the latitude of the tracker (geodetic)
- h be the altitude of the tracker above the reference spheroid
- a be the body semi-major axis
- b be the body semi-minor axis

$$\text{Let } c = \sqrt{\cos^2 \phi + \frac{b^2}{a^2} \sin^2 \phi}$$

Then

$$\text{SN}(1) = \left(\frac{a}{c} + h \right) \cos \theta \cos \phi$$

$$\text{SN}(2) = \left(\frac{a}{c} + h \right) \sin \theta \cos \phi$$

$$\text{SN}(3) = \left(\frac{b^2}{ac} + h \right) \sin \phi$$

DSTAT -2

Transformation

The columns of the topocentric to body-fixed transformation matrix (stored by columns in SN(4) to SN(12)) are the local unit North, East, and Down vectors:

$$\begin{array}{ccc} & \text{N} & \text{E} & \text{D} \\ \left[\begin{array}{ccc} -\cos \theta \sin \phi & -\sin \theta & -\cos \theta \cos \phi \\ -\sin \theta \sin \phi & \cos \theta & -\sin \theta \cos \phi \\ \cos \phi & 0 & -\sin \phi \end{array} \right] \end{array}$$

DSTAT-3

```

$IBFTC MC13DW XR3,M94,NODD,LIST
SUBROUTINE DSTAT(A,B,DTR,STA,SN)
C FINDS STATION POSITION VECTORS AND TRANSFORMATIONS
DOUBLE PRECISION A, B, STA(3), SN(12)
1, SL, CL, ST, CT, C, DTR
DOUBLE PRECISION DSIN,DCOS,DSQRT
SL=DSIN(STA(1)*DTR)
CL=DCOS(STA(1)*DTR)
ST=DSIN(STA(2)*DTR)
CT=DCOS(STA(2)*DTR)
C=DSQRT(CL*CL+B*B*SL*SL/(A*A))
SN(4)=-SL*CT
SN(5)=-SL*ST
SN(6)=CL
SN(7)=-ST
SN(8)=CT
SN(9)=0.D0
SN(10)=-CL*CT
SN(11)=-CL*ST
SN(12)=-SL
SN(3) = STA(3)/1.D3
SN(1)=- (A/C+SN(3))*SN(10)
SN(2)=- (A/C+SN(3))*SN(11)
SN(3)=- (B*B/(A*C)+SN(3))*SN(12)
RETURN
END
STAT0001
STAT0002
STAT0003
STAT0004
STAT0005
STAT0006
STAT0007
STAT0008
STAT0009
STAT0010
STAT0011
STAT0012
STAT0013
STAT0014
STAT0015
STAT0016
STAT0017
STAT0018
STAT0019
STAT0020
STAT0021
STAT0022
STAT0023
STAT0024

```

Subroutine: DSTEST

Purpose: To decode data and convert units for DSIF(JPL)
data.

Calling Sequence: CALL DSTEST(IERR)

I/φ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
φ	IERR				Error flag. Set = 2 by DSTEST if total number of station names exceeds 20. Otherwise set = 0.

Common storages used: /TRKCOM/, /OUTCOM/, /TSTCOM/, /DATCOM/, /SUMCOM/

Subroutines required: DATINP, POLYFT

DSTEST-1

Discussion:

Subroutine DSTEST is presently a dummy subroutine. If it is called, it writes out an error message and returns control to the main program for processing of the next data tape.

Subroutine DSTEST was intended to parallel the operation of the similar subroutines CBTEST, GRTEST, and SBTEST, but could not be written because neither descriptions nor samples of DSIF (JPL) data were made available.

DSTEST-2

```
SIBFTC MC134V M94,NODD,XR3
CMC134V DSIF RAW DATA PROCESSOR
SUBROUTINE DSTEST(IERR)
601 FORMAT(48H0*** DSIF DATA PROCESSING NOT PERMITTED BECAUSE/
16X,41HFORMATTING INFORMATION WAS NOT AVAILABLE./
26X,34HPROCESSING OF THIS TAPE ABANDONED./
36X,38HPROGRAM PROCEEDS TO NEXT TAPE, IF ANY.)
IERR=0
WRITE(6,601)
RETURN
END
```

```
DSTS0001
DSTS0002
DSTS0003
DSTS0004
DSTS0005
DSTS0006
DSTS0007
DSTS0008
DSTS0009
DSTS0010
```

Subroutine: DTRANP

Purpose: To transform the covariance matrix from one inertial frame to another (given the transformation matrix).

Calling Sequence: CALL DTRANP(P, NR, T, S, Q, KK)

Input and Output

I/∅	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	P	d(NR, NR)			Input covariance matrix
I	NR				Dimension of P
I	T	d(3, 3)			Position transformation matrix
I	S	d(3, 3)			Velocity transformation matrix
∅	Q	d(6, 6)			Output covariance matrix
I	KK				Option key; see description following.

Common storages used: 54 cells.

Subroutine required: DMVTRN

DTRANP- 1

DTRANP Usage

Consider the upper left 6×6 of P as being partitioned:

$$\begin{array}{c} 3 \\ 3 \end{array} \left[\begin{array}{c|c} 3 & 3 \\ \hline p_1 & p_2 \\ \hline p_3 & p_4 \end{array} \right]$$

If $KK = 0$

$$Q = \begin{bmatrix} T & 0 \\ 0 & S \end{bmatrix} \begin{bmatrix} p_1 & p_2 \\ p_3 & p_4 \end{bmatrix} \begin{bmatrix} T^T & 0 \\ 0 & S^T \end{bmatrix}$$

If $KK = 1$

$$Q = \begin{bmatrix} T p_1 T^T & Q' \\ Q' & S p_4 S^T \end{bmatrix}$$

where Q' is left as it was on input.

```

$IBFTC MC13DJ XR3,M94,NODD,LIST
SUBROUTINE DTRANP(P, NR, T, S, Q, KK)
C TRANSFORMS COVARIANCE MATRIX FROM ONE INERTIAL FRAME TO ANOTHER
DOUBLE PRECISION T(3,3), P(NR,NR), Q(6,6), S(3,3)
COMMON A,B,D
DOUBLE PRECISION A(3,3), B(3,3), D(3,3)
C Q = ( T 0 ) (P1 P2) (TT 0) WHERE P1, P2, P3, P4 ARE
C ( 0 S ) (P3 P4) (0 ST) 3X3 MATRICES, UPPER LEFT OF P
C KK=0 GIVES FULL 6X6 IN Q
C KK=1 GIVES T P1 TT AND S P4 ST IN Q1, Q4
DO 1 I=1,3
DO 1 J=1,3
A(I,J)=T(J,I)
1 B(I,J)=S(J,I)
DO 2 I=1,3
2 CALL DMVTRN(T,P(1,I),D(1,I),1,1)
DO 3 I=1,3
3 CALL DMVTRN(D,A(1,I),Q(1,I),1,1)
DO 4 I=1,3
4 CALL DMVTRN(S,P(4,I+3),D(1,I),1,1)
DO 5 I=1,3
5 CALL DMVTRN(D,B(1,I),Q(4,I+3),1,1)
IF(KK.GT.0) GO TO 9
DO 6 I=1,3
6 CALL DMVTRN(T,P(1,I+3),D(1,I),1,1)
DO 7 I=1,3
7 CALL DMVTRN(D,B(1,I),Q(1,I+3),1,1)
DO 8 I=4,6
DO 8 J=1,3
8 Q(I,J)=Q(J,I)
9 RETURN
END
TRNP0001
TRNP0002
TRNP0003
TRNP0004
TRNP0005
TRNP0006
TRNP0007
TRNP0008
TRNP0009
TRNP0010
TRNP0011
TRNP0012
TRNP0013
TRNP0014
TRNP0015
TRNP0016
TRNP0017
TRNP0018
TRNP0019
TRNP0020
TRNP0021
TRNP0022
TRNP0023
TRNP0024
TRNP0025
TRNP0026
TRNP0027
TRNP0028
TRNP0029
TRNP0030

```

Subroutine: DTRDB

Purpose: To compute either the Darboux or the local tangent plane transformation.

Calling Sequence: CALL DTRDB(R, V, T, K)

Input and Output

I/∅	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	R	d(3)			Position, body centered coordinates
I	V	d(3)			Velocity, body centered coordinates
∅	T	d(3,3)			Transformation matrix
I	K				K = 1, T is Darboux K = 2, T is local tangent plane.

Common storages used: None

Subroutines required: DCR~~∅~~SS, DN~~∅~~RM

DTRDB-1

DTRDB Equations:

1. Darboux

The columns of T are the vectors

$VX(RXV)$, $(RXV) \times [VX(RXV)]$, RXV : all normalized

2. Local tangent plane

R , $(RXV)XR$, RXV : all normalized.

DTRDB-2

Subroutine: DTRDB

Purpose: To compute either the Darboux or the local tangent plane transformation.

Calling Sequence: CALL DTRDB(R, V, T, K)

Input and Output

I/∅	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	R	d(3)			Position, body centered coordinates
I	V	d(3)			Velocity, body centered coordinates
∅	T	d(3,3)			Transformation matrix
I	K				K = 1, T is Darboux K = 2, T is local tangent plane.

Common storages used: None

Subroutines required: DCR∅SS, DN∅RM

DTRDB-1

DTRDB Equations:

1. Darboux

The columns of T are the vectors

$VX(RXV)$, $(RXV) \times [VX(RXV)]$, RXV : all normalized

2. Local tangent plane

R , $(RXV)XR$, RXV : all normalized.

DTRDB-2

\$IBFTC MC13DF XR3,M94,NODD,LIST	
SUBROUTINE DTRDB(R,V,T,K)	TRDB0001
C COMPUTES DARBOUX OR LOCAL TANGENT PLANE TRANSFORMATION	TRDB0002
DOUBLE PRECISION T(3,3), R(3), V(3), A,	TRDB0003
DOUBLE PRECISION DNORM	TRDB0004
C K=1 GIVES DARBOUX	TRDB0005
C K=2 GIVES LOCAL TP	TRDB0006
CALL DCROSS(R,V,T(1,3))	TRDB0007
IF(K.EQ.1) GO TO 2	TRDB0008
DO 1 I=1,3	TRDB0009
1 T(I,1)=R(I)	TRDB0010
GO TO 3	TRDB0011
2 CONTINUE	TRDB0012
CALL DCROSS(V,T(1,3),T(1,1))	TRDB0013
3 A=DNORM(T(1,1))	TRDB0014
B=DNORM(T(1,3))	TRDB0015
DO 4 I=1,3	TRDB0016
T(I,3)=T(I,3)/B	TRDB0017
4 T(I,1)=T(I,1)/A	TRDB0018
CALL DCROSS(T(1,3),T(1,1),T(1,2))	TRDB0019
RETURN	TRDB0020
END	

Subroutine: DVNORM

Purpose: To normalize a vector.

Calling Sequence: X = DVNORM (Z, W)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	Z	d(3)			Input vector.
Ø	W	d(3)			Unit vector along Z.
Ø	DVNORM	d			Magnitude of Z.

Common storages used: None

Subroutines required: DDØT

DVNORM-1

```
$IBFTC MC132H XR3,M94,NODD,LIST
DOUBLE PRECISION FUNCTION DVNORM (Z,W)
DOUBLE PRECISION      Z(3),W(3),SCL,DDOT,DSQRT
1 SCL = DSQRT(DDOT(Z,Z))
DO 2 I=1,3
2 W(I) = Z(I)/SCL
DVNORM = SCL
RETURN
END
```

```
DVNM0001
DVNM0002
DVNM0003
DVNM0004
DVNM0005
DVNM0006
DVNM0007
```

Subroutine: EGRESS

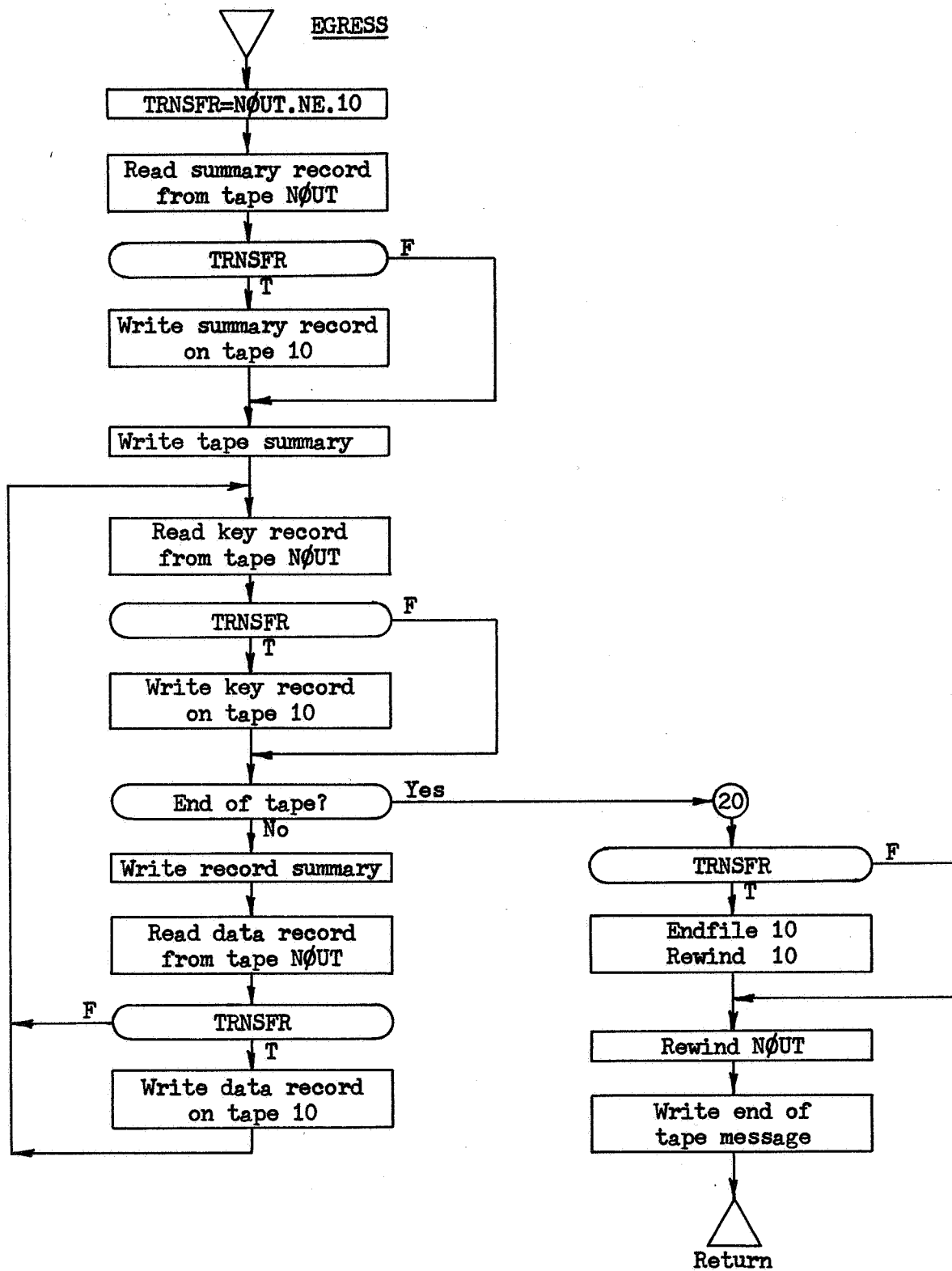
Purpose: To transfer the final edited data to tape 10 if necessary,
and to write the output summary.

Calling Sequence: CALL EGRESS

Common storages used: /MIXCOM/, /SUMCOM/, /OUTCOM/, /MESC0M/

Subroutines required: DAT0UP

EGRESS-1



EGRESS-2

```

$IBFTC MC134U M94,NODD,XR3
CMC134U WRITES SUMMARY AND INSURES OUTPUT TO BE ON TAPE 10 EGRS0001
SUBROUTINE EGRESS EGRS0002
COMMON /MIXCOM/IMIX(4) EGRS0003
EQUIVALENCE (IMIX(1),LONNIN), (IMIX(2),LON12) EGRS0004
1, (IMIX(3),NIN), (IMIX(4),NOUT) EGRS0005
C EGRS0006
COMMON /SUMCOM/SUMM(56) EGRS0007
DOUBLE PRECISION TSTART, TSTOP EGRS0008
DIMENSION HEADER(11), STIMNX(20) EGRS0009
REAL NAMSTA(20) EGRS0010
EQUIVALENCE (SUMM(1),HEADER), (SUMM(12),NUMSTA), (SUMM(13),NAMSTA) EGRS0011
1, (SUMM(33),TSTART), (SUMM(35),TSTOP), (SUMM(37),STIMNX) EGRS0012
C EGRS0013
COMMON /OUTCOM/BUFF(40) EGRS0014
DIMENSION SSD(4) EGRS0015
DOUBLE PRECISION TFIRST, TLAST EGRS0016
EQUIVALENCE (BUFF(1),NREC), (BUFF(2),NEOT), (BUFF(3),NAMOBS) EGRS0017
1, (BUFF(4),NAMTRA), (BUFF(6),NPTS), (BUFF(7),KONT) EGRS0018
2, (BUFF(8),MTYPE), (BUFF(15),TFIRST), (BUFF(17),TLAST) EGRS0019
3, (BUFF(33),NB1), (BUFF(34),NB2), (BUFF(35),NB3) EGRS0020
4, (BUFF(36),NB4), (BUFF(37),SSD) EGRS0021
C EGRS0022
COMMON /MESCOM/DATA(510) EGRS0023
C EGRS0024
DIMENSION TYPE(4), DUM(3) EGRS0025
LOGICAL TRNSFR EGRS0026
C EGRS0027
601 FORMAT(1H1,46X,26H*** EDITED DATA TAPE ***/24X,7HHEADER,11A6) EGRS0028
602 FORMAT(19H0FIRST DATA ON TAPE,D23.16,20H SECONDS FROM 1950.0) EGRS0029
603 FORMAT(19H0 LAST DATA ON TAPE,D23.16,20H SECONDS FROM 1950.0) EGRS0030
604 FORMAT(1H0,51X,16HSTATIONS ON TAPE/49X,4HNAME,5X,13HFIRST ON TIME/ EGRS0031
1/(48X,A6,E18.8)) EGRS0032
605 FORMAT(//////24H *** RECORD PAIR NUMBER,I5//75H RECEIVING STATION EGRS0033
1 TRANSMITTING STATION MEASUREMENT TYPE CONTINUATION/7X,A6,15 EGRS0034
2X,A6,15X,A6,12X,A3//12H DATA BEGINS,D24.16,20H SECONDS FROM 1950.0 EGRS0035
3) EGRS0036
606 FORMAT(12H0 DATA ENDS ,D24.16,20H SECONDS FROM 1950.0) EGRS0037
607 FORMAT(1H0,3X,5HTOTAL,25X,10HBAD POINTS/12H TIME POINTS,9X,7HANGLE EGRS0038
1 1,3X,7HANGLE 2,3X,5HRANGE,3X,7HDOPPLER/I7,17X,I2,8X,I2,6X,I2,7X, I EGRS0039
22//30X,19HSTANDARD DEVIATIONS/5X,7HANGLE 1,13X,7HANGLE 2,14X,5HRAN EGRS0040
3GE,14X,7HDOPPLER/E16.8,3E20.8) EGRS0041
608 FORMAT(//44X,32H *** END OF EDITED DATA TAPE ***) EGRS0042
DATA TYPE/6HC-BAND,6HGODDRD,6HHS-BAND,6H DSIF / EGRS0043
DATA QYES,QNO/3HYES,3H NO/ EGRS0044
TRNSFR=NOUT.NE.10 EGRS0045
C READ TAPE SUMMARY RECORD EGRS0046
READ(NOUT) SUMM EGRS0047
C IF NOUT .NE. 10 WRITE IT ON 10 EGRS0048
IF(TRNSFR) WRITE(10) SUMM EGRS0049
C WRITE OUT SUMMARY RECORD DATA ON SYSOU1 EGRS0050
WRITE(6,601) HEADER EGRS0051
WRITE(6,602) TSTART EGRS0052
CALL DATOUP(TSTART,DUM,0) EGRS0053
WRITE(6,603) TSTOP EGRS0054
CALL DATOUP(TSTOP,DUM,0) EGRS0055
WRITE(6,604) (NAMSTA(I),STIMNX(I),I=1,NUMSTA) EGRS0056
C READ A KEY RECORD FROM NOUT EGRS0057
10 READ(NOUT) BUFF EGRS0058
C IF NOUT .NE. 10 WRITE IT ON 10 EGRS0059
IF(TRNSFR) WRITE(10) BUFF EGRS0060
C TEST FOR END OF TAPE EGRS0061
IF(NEOT.EQ.1) GO TO 20 EGRS0062
QOUT=QNO EGRS0063
IF(KONT.EQ.1) QOUT=QYES EGRS0064
C WRITE SUMMARY OF KEY RECORD ON SYSOU1 EGRS0065
WRITE(6,605) NREC,NAMOBS,NAMTRA,TYPE(MTYPE),QOUT,TFIRST EGRS0066
CALL DATOUP(TFIRST,DUM,0) EGRS0067
WRITE(6,606) TLAST EGRS0068
CALL DATOUP(TLAST,DUM,0) EGRS0069
WRITE(6,607) NPTS,NB1,NB2,NB3,NB4,SSD EGRS0070
C READ DATA RECORD EGRS0071
READ(NOUT) DATA EGRS0072
C IF NOUT .NE. 10 WRITE IT ON 10 EGRS0073
IF(TRNSFR) WRITE(10) DATA EGRS0074

```

GO TO 10
20 CONTINUE
IF(TRANSFR) ENDFILE 40
IF(TRANSFR) REWIND 40
REWIND NOUT
WRITE(6,608)
RETURN
END

EGRS0075
EGRS0076
EGRS0077
EGRS0078
EGRS0079
EGRS0080
EGRS0081
EGRS0082

Subroutine: ENCKED

Purpose: To compute the inverse-square perturbing acceleration due to a central body (the so-called "Encke acceleration") in double precision.

Calling Sequence: CALL ENCKED(U,R,D,A)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	U	d	μ	km ³ /sec ²	Central body gravitational constant.
I	R	d(3)	R _o	km	Reference conic Position vector
I	D	d(3)	δR	km	Displacement from the reference conic position.
Ø	A	d(3)	$\ddot{\delta R}$	km/sec ²	Encke acceleration.

Common storages used: 6 cells

Subroutines required: DDØT

ENCKED-1

Description:

In using Encke's method for integration of trajectories, we set

$$R = R_0 + \delta R$$

where R is the position vector, R_0 is the position vector on a reference conic, and δR is the displacement from the conic to the true position.

$$\ddot{R} = -\frac{\mu}{r^3} R$$

$$\ddot{R}_0 = -\frac{\mu}{r_0^3} R_0$$

where $r = |R|$, etc., then

$$\delta \ddot{R} = -\frac{\mu}{r^3} R + \frac{\mu}{r_0^3} R_0$$

where $\delta \ddot{R}$ is the acceleration of R relative to R_0 due to the inverse-square attraction of the central body. The terms in $\delta \ddot{R}$ are large compared to $\delta \ddot{R}$, and to save significance, we compute $\delta \ddot{R}$ from the expansion

$$\delta \ddot{R} = \frac{\mu b}{r_0^3} (cR_0 - \delta R)$$

where

$$q = (2R_0 + \delta R) \cdot \delta R / r_0^2$$

$$p = -1 + (1 + q)^3 = q(3 + 3q + q^2)$$

$$b = 1/\sqrt{1 + p}$$

$$c = p/(1 + \sqrt{1 + p})$$

ENCKED-2

```

$IBFTC MC1320 XR3,M94,NODD,LIST
SUBROUTINE ENCKED (U,R,D,A)
ENCKE ACCELERATION
C DOUBLE PRECISION A(3),D(3),R(3),U
1 ,C,DDOT,DSQRT
COMMON C(3)
C
1 C(1) = DDOT(R,R)
C(2) = DDOT(D,D)
C(3) = (2.D0*DDOT(R,D)+C(2))/C(1)
C(2) = C(3)*(3.D0+C(3)*(3.D0+C(3)))
C(3) = 1.D0+C(2)
C(3) = DSQRT(C(3))
C(1) = C(3)*(C(1)*DSQRT(C(1)))/U
C(3) = 1.D0+C(3)
C(2) = C(2)/C(3)
DO 2 I=1,3
2 A(I) = (C(2)*R(I)-D(I))/C(1)
999 RETURN
END

```

```

ENKE0001
ENKE0002
ENKE0003
ENKE0004
ENKE0005
ENKE0006
ENKE0007
ENKE0008
ENKE0009
ENKE0010
ENKE0011
ENKE0012
ENKE0013
ENKE0014
ENKE0015
ENKE0016
ENKE0017
ENKE0018

```

Subroutine: ERRØUT (Dummy)

Purpose: Provides a dummy ERRØUT in FORTRAN IV to satisfy the needs of several subroutines that call ERRØUT. (previously, a MAP-coded error trace routine).

Calling Sequence: CALL ERRØUT (List ignored)

Common storages used: None

Subroutines required: None

ERRØUT-1

```
$IBFTC MC134A M94,NODD,NOREF,XR3
  SUBROUTINE ERR0UT
C      THIS IS A DUMMY REPLACEMENT FOR THE MAP-CODED
C      ERROR TRACE ROUTINE ERR0UT
  WRITE(6,601)
 601 FORMAT(29H0ERROR STOP FROM DUMMY ERR0UT)
  STOP
  END
```

```
ERDM0001
ERDM0002
ERDM0003
ERDM0004
ERDM0005
ERDM0006
ERDM0007
```

Subroutine: ESTMAT

Purpose: Controls accumulation of differential corrections to a state estimate using the data from a specified tracking station.

Calling Sequence: CALL ESTMAT (LTRAJ)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
IØ	LTRAJ	ℓ			.TRUE. except on initial entry and final return.

Common storages used: //270 cells, /DATCOM/, /DCPCOM/, /DCRCOM/, /EDTCOM/, /ESTCOM/, /ES1COM/, /SBFCOM/

Subroutines required: CBDATP, DDØT, DEHA, DEQTR, DGTRN, DGTSN, DMPly, DMVTRN, DSDATP, ESTØUT, GRDATP, SBDATP

ESTMAT-1

Method

ESTMAT uses the modified Kalman filter described in Reference 1, Appendix B, to compute differential corrections to an a priori state estimate and its covariance matrix using the data from a single tracking station. Let

$$Z = \begin{bmatrix} X(t) \\ U \\ V \end{bmatrix}$$

denote the true state, where

$X(t)$ = vehicle state,

U = a set of parameters affecting the motion, and

V = a set of parameters affecting the measurements.

We assume that a "nominal" state, \bar{Z} , is stored and define the deviations

$$\bar{Z} = Z_0 + \bar{z}$$

$$\hat{Z} = Z_0 + \hat{z} = \text{current state estimate}$$

where Z_0 is the a priori estimate of state.

Interpolation on the Nominal Trajectory

TRAJDP integrates a nominal trajectory, $\bar{X}(t_v)$, from the initial conditions stored in ESTCØM. The nominal trajectory, transition matrix, parameter sensitivity matrix, Earth nutations, and down-leg transmission delay time are provided to ESTMAT as a set of interpolation coefficients in SBFCØM. Let

ESTMAT-2

$R(t_v)$ = vehicle position on the nominal trajectory (3x1),
 $\phi(t_v; t_{v1})$ = state transition matrix, integrated along the
 nominal trajectory (3x6),
 $\phi_u(t_v; t_{v1})$ = parameter sensitivity matrix, integrated along the
 $\delta\psi, \delta\epsilon$ = nutations in longitude, obliquity,
 $\delta t(t_r)$ = down-leg transmission delay
 = $t_r - t_v(t_r)$,

and let A be the vector

$$A = \begin{bmatrix} R(t_v) \\ \phi(t_v; t_{v1}) \\ \phi_u(t_v; t_{v1}) \\ a_n \\ a_{n+1} \\ a_{n+2} \end{bmatrix} \quad \dot{A} = \begin{bmatrix} \dot{R}(t_v) \\ \dot{\phi}(t_v; t_{v1}) \\ \dot{\phi}_u(t_v; t_{v1}) \\ \dot{\delta\psi} \\ \dot{\delta\epsilon} \\ \dot{\delta t}(t_r) \end{bmatrix}$$

where the components of ϕ, ϕ_u are stored in column order in sequential components of A.

SBFCOM contains

TBF = t_o = the initial time for the interpolation interval,
 TFF = t_f = the final time for the interpolation interval,
 = $t_o + 7 h$,
 HBF = h = the interval size used in constructing the
 interpolation tables,
 RBF = $A(t_o)$
 VBF = $\{\alpha_{ij}\}$ = interpolation coefficients for $\dot{A}(t)$.

For any t, we set

$$d = (t - t_o)/h$$

ESTMAT-3

and compute

$$\begin{aligned} \dot{a}_i &= \sum_{j=1}^8 \alpha_{ij} d^{j-1}/(j-1)! \\ a_i &= a_i(t_0) + \sum_{j=1}^8 \alpha_{ij} d^j/j! \end{aligned}$$

The measurement data is given at known values of reception time, t_{Sr} . In ephemeris time,

$$\begin{aligned} t_{Ur} &= t_{Sr} + \gamma_1 + \gamma_2 t_{Sr} \\ t_r &= t_{Ur} + \gamma_3 + \gamma_4 t_{Ur} \end{aligned}$$

We first interpolate for $\delta t(t_r)$ and compute

$$t_v = t_r - \delta t(t_r)$$

The remaining components of A and \dot{A} are then computed as functions of t_v .

Computation of Residuals

Let

$$y = G(X(t), V, t) + q$$

be the measurement value, where q is a random error. We compute the observable on the nominal

$$y = G(\bar{X}(t), \bar{V}, t)$$

and the residual from the nominal, $y - \bar{y}$, using the measurement model described in Reference 1, Appendix C. Since

$$\hat{z} = (\bar{z} - z) + \hat{z}$$

ESTMAT-4

the residual from the current estimate is

$$y - \hat{y} = y - \bar{y} - H\hat{z} + H\bar{z}$$

where

$$H = \frac{\partial y}{\partial z}$$

Differential Corrections

Let

$$P = E((Z - \hat{Z})(Z - \hat{Z})^T)$$

$$Q = E(qq^T)$$

$$E(q) = 0$$

$$E(qZ^T) = 0$$

The differential corrections are computed from

$$\hat{z} \leftarrow \hat{z} + J(PH^T + \epsilon H^T)(HPH^T + Q)^{-1}(y - \hat{y})$$

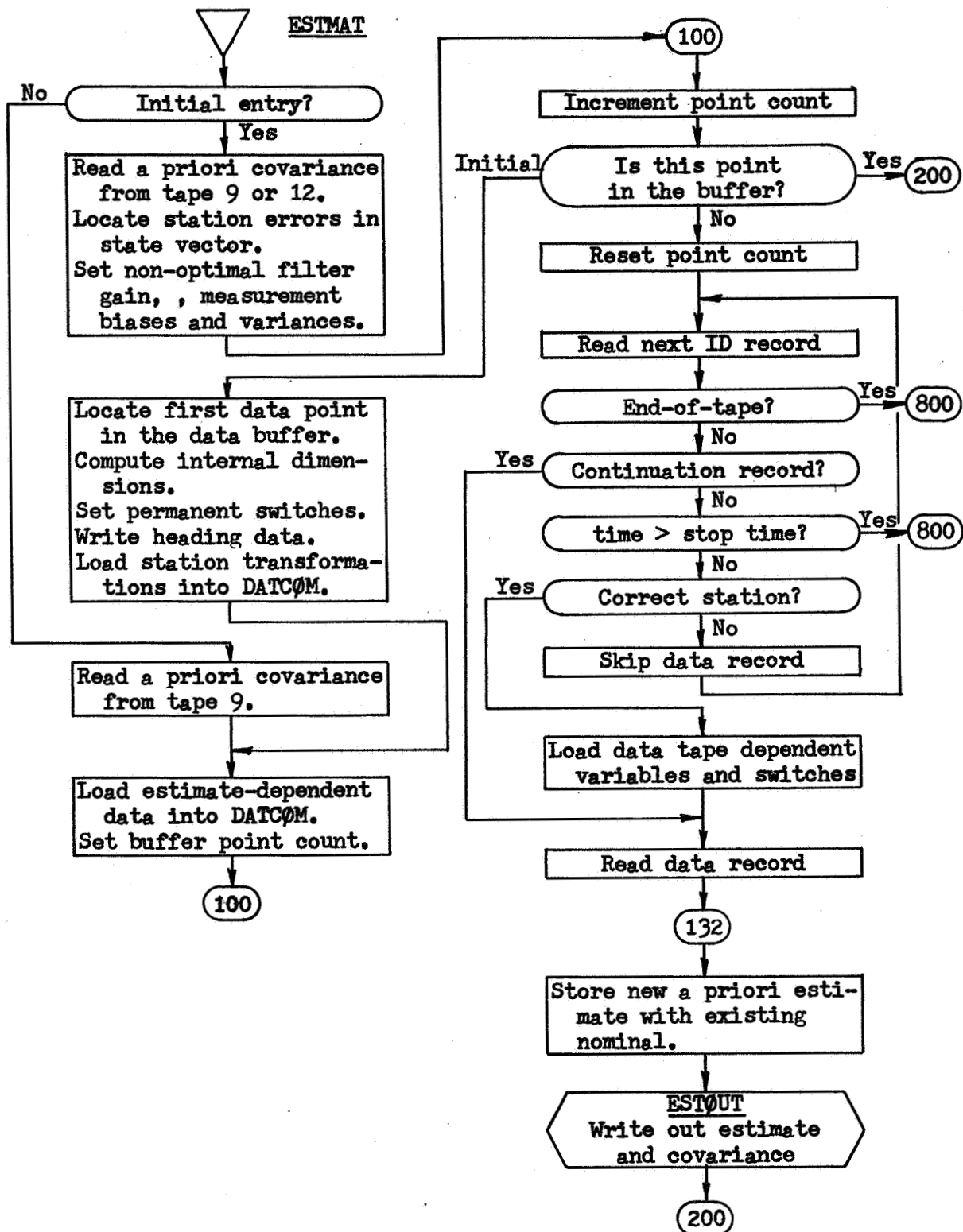
$$P \leftarrow P - J(PH^T + \epsilon H^T)(HPH^T + Q)^{-1}(PH^T + \epsilon H^T)^T$$

where J is a diagonal matrix with

$$J_{ii} = 1 \text{ if } Z_i \text{ is to be estimated}$$

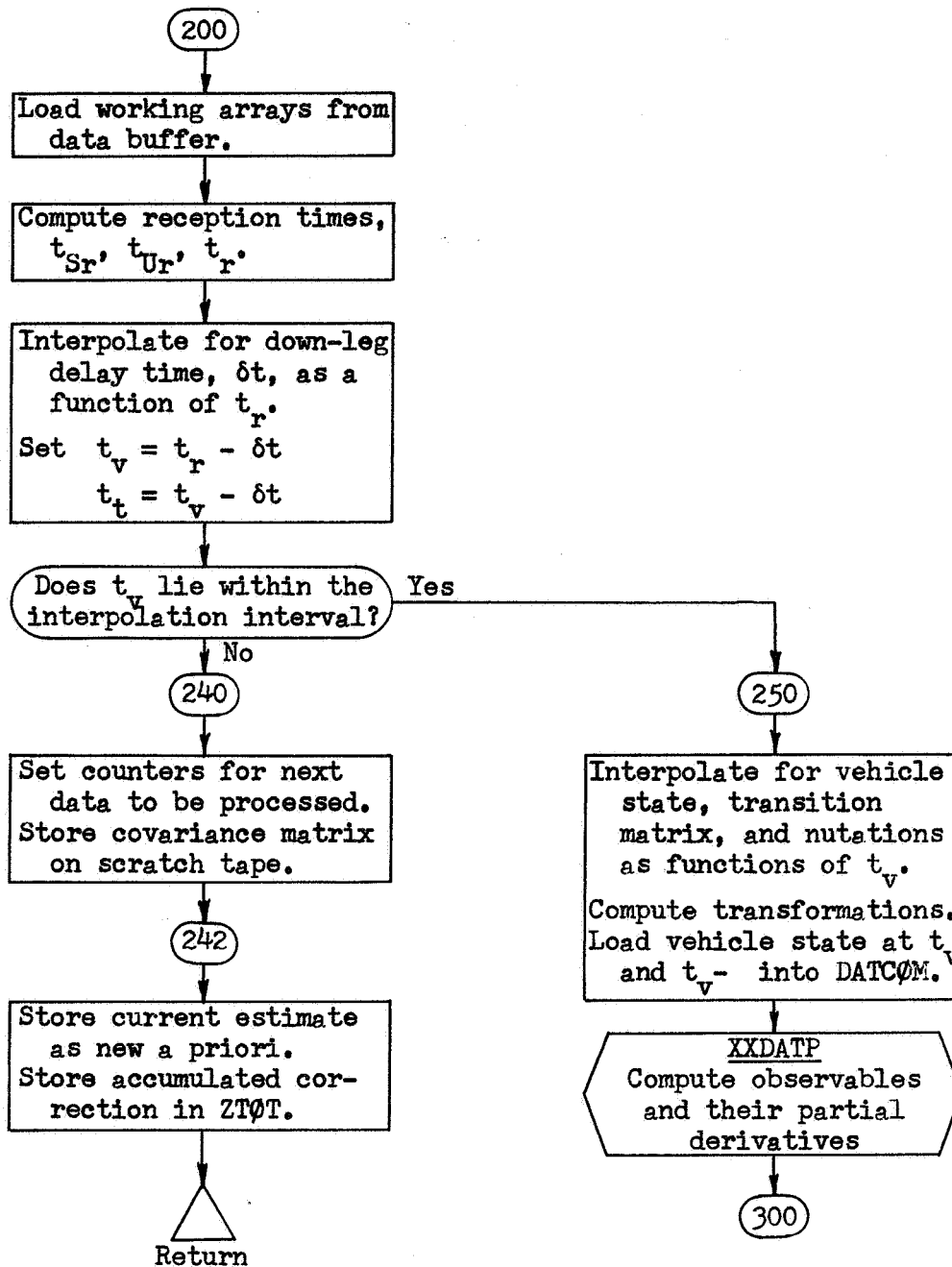
$$J_{ii} = 0 \text{ if } Z_i \text{ is to be included as uncertain, but not solved for.}$$

ESTMAT-5



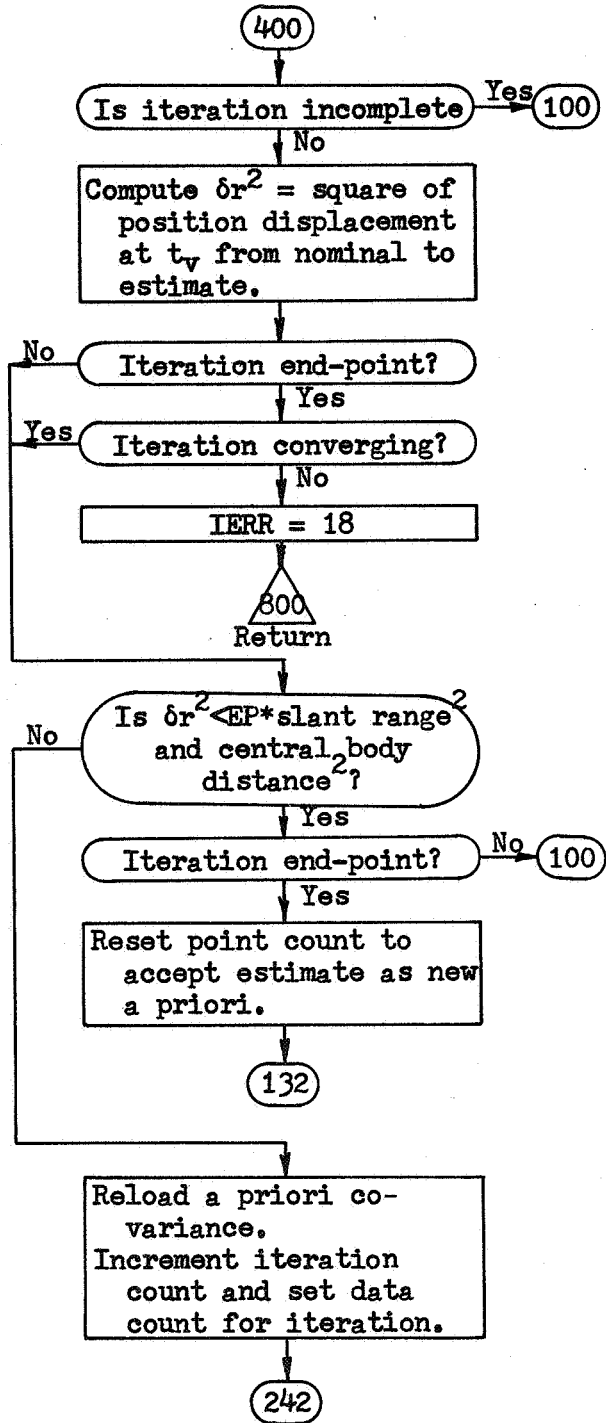
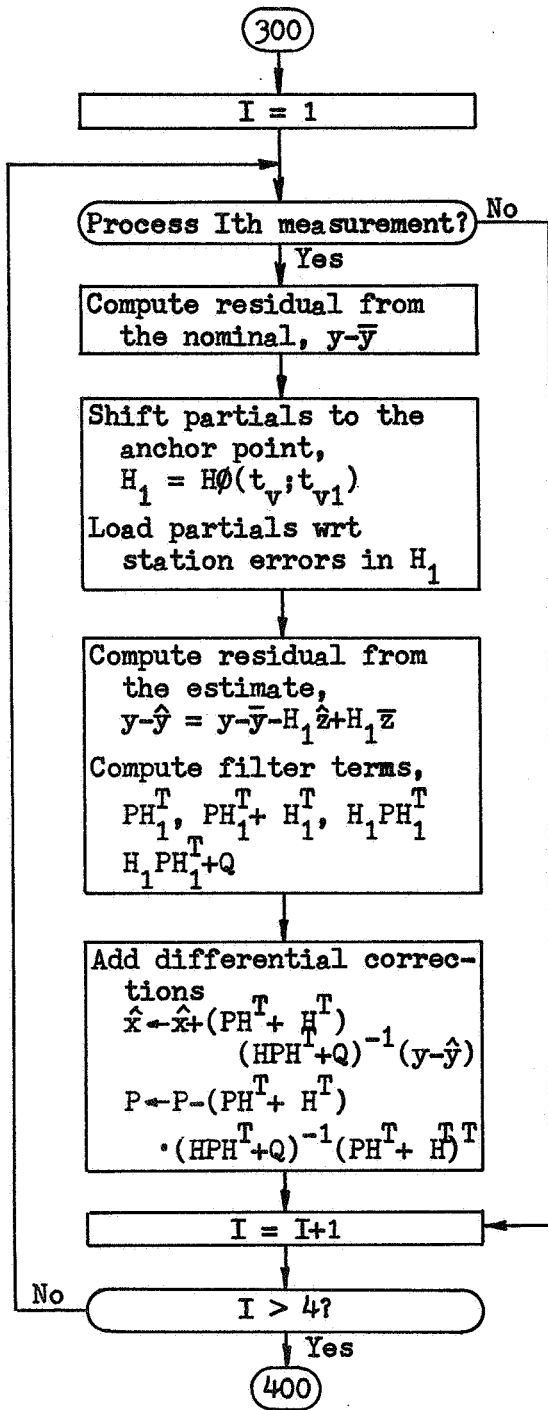
Initialization and
Data Input Control

ESTMAT-6



Computation of Observables and Partial Derivatives

ESTMAT-7



ESTMAT-8

```

$IBFTC MC133J XR3,M94,NODD,LIST
SUBROUTINE ESTMAT (LTRAJ)
CONTROL ROUTINE FOR DIFFERENTIAL CORRECTION OF ESTIMATE
ESTM0001
C
C LOGICAL LTRAJ ESTM0002
DOUBLE PRECISION DDOT ,DCOS ,DSIN ESTM0003
C ESTM0004
COMMON /DCPCOM/CDCP(900) ESTM0005
DOUBLE PRECISION CBRAN2 ,EPSDC ,SN(13,20) ESTM0006
1 ,CON(8) ,STIMR (2) ESTM0007
DIMENSION IFSKIP(4) ,IPROC(5) ,MSKIP(4,4) ,NPROC(22) ESTM0008
EQUIVALENCE (CDCP(811),EPSDC ) ,(CDCP(700),NPEND ) ESTM0009
1 ,(CDCP(111),IERR ) ,(CDCP(731),NPROC ) ESTM0010
2 ,(CDCP(756),IPROC ) ,(CDCP(143),SN ) ESTM0011
3 ,(CDCP(899),CBRAN2) ,(CDCP(881),MSKIP ) ,(CDCP(781),STIMR ) ESTM0012
4 ,(CDCP( 1),CON ) ,(CDCP(702),NEMPS ) ,(CDCP(110),YTEST ) ESTM0013
EQUIVALENCE (IPROC( 5),IFTSTP ) ,(IPROC( 2),ITRSTA) ESTM0014
1 ,(IPROC( 3),INDSTA) ESTM0015
EQUIVALENCE (NPROC(18),IFBADQ) ,(NPROC(10),NPREST) ESTM0016
1 ,(NPROC(19),IFOUTL ) ,(NPROC( 3),NPRSTA) ESTM0017
2 ,(NPROC(14),IFSKIP ) ,(NPROC(13),NPSKIP) ESTM0018
C ESTM0019
COMMON /EDTCOM/INDDAT(40),BUFDAT(85,6) ESTM0020
DOUBLE PRECISION DATIND(20),ONTIME ESTM0021
EQUIVALENCE (INDDAT( 7),KONT ) ,(INDDAT( 6),NPTS ) ESTM0022
1 ,(INDDAT(12),KTAU ) ,(INDDAT( 3),NRSTA ) ESTM0023
2 ,(INDDAT( 8),MTYPE ) ,(INDDAT( 4),NTSTA ) ESTM0024
3 ,(INDDAT( 1),DATIND) ,(INDDAT( 2),NEOT ) ,(INDDAT(13),ONTIME) ESTM0025
C ESTM0026
COMMON /ESTCOM/CEST(804) ESTM0027
DOUBLE PRECISION DELDAN(2) ,SE(14,20) ,SPCDAN(6) ESTM0028
EQUIVALENCE (CEST( 97),DELDAN) ,(CEST(245),SE ) ESTM0029
1 ,(CEST( 27),KOUNTN) ,(CEST( 49),SPCDAN) ESTM0030
C ESTM0031
COMMON /ESLCOM/PNEW(30,30),STNAMN(30),TRAKER(30) ESTM0032
1 ,ITRETN (30),ILCCN (30),KLCCN (54) ESTM0033
2 ,NSN (12,20) ESTM0034
DOUBLE PRECISION PNEW ESTM0035
DIMENSION DEST(2214) ESTM0036
EQUIVALENCE (PNEW,DEST) ESTM0037
C ESTM0038
COMMON /SBFCOM/CSBF(12),RBF(45),VBF(45,8) ESTM0039
DOUBLE PRECISION RBF ,VBF ,HBF ,TBF ,TFB ESTM0040
EQUIVALENCE (CSBF( 6),ITER ) ,(CSBF( 2),NTP ) ESTM0041
1 ,(CSBF( 3),NPT1 ) ,(CSBF( 9),TBF ) ESTM0042
2 ,(CSBF( 7),HBF ) ,(CSBF( 4),NPT2 ) ,(CSBF(11),TFB ) ESTM0043
C ESTM0044
COMMON /DCRCOM/ZBAR(30),ZHAT(30),ZTOT(30),DCSAVE(74) ESTM0045
DOUBLE PRECISION ZBAR ,ZHAT ,ZTOT ESTM0046
DOUBLE PRECISION DPREV ,OBIAS(4) ,TFIRST ESTM0047
1 ,EPSK(4) ,Q2 (4) ,YMOD(4) ESTM0048
LOGICAL LSKIP(4) ,LSTA ,KOUTL ESTM0049
DIMENSION NASGN(9) ,LABEL(12) ,SYMBOL(4) ESTM0050
EQUIVALENCE (DCSAVE(39),LABEL ) ,(DCSAVE( 1),OBIAS ) ESTM0051
1 ,(DCSAVE(55),LSKIP ) ,(DCSAVE(25),Q2 ) ESTM0052
2 ,(DCSAVE(33),DPREV ) ,(DCSAVE(59),LSTA ) ,(DCSAVE(37),STPTIM) ESTM0053
3 ,(DCSAVE(17),EPSK ) ,(DCSAVE(66),NASGN ) ,(DCSAVE(51),SYMBOL) ESTM0054
4 ,(DCSAVE(64),LSTA ) ,(DCSAVE(61),NDELTA ) ,(DCSAVE(35),TFIRST) ESTM0055
5 ,(DCSAVE(65),LSTA ) ,(DCSAVE(63),NHU ) ,(DCSAVE(38),TREF ) ESTM0056
6 ,(DCSAVE(60),KOUTL ) ,(DCSAVE(62),NINT ) ,(DCSAVE( 9),YMOD ) ESTM0057
EQUIVALENCE (NASGN( 2),N129 ) ,(NASGN( 6),N204 ) ESTM0058
1 ,(NASGN( 3),N130 ) ,(NASGN( 7),N290 ) ESTM0059
2 ,(NASGN( 4),N200 ) ,(NASGN( 8),N300 ) ESTM0060
3 ,(NASGN( 1),N122 ) ,(NASGN( 5),N203 ) ,(NASGN( 9),N302 ) ESTM0061
C ESTM0062
COMMON /CATCOM/CDAT(299) ESTM0063
DIMENSION IDATA(7) ESTM0064
DOUBLE PRECISION BIAS (2) ,OMEGA ,TAU ,TT2BT(9) ESTM0065
1 ,FTR ,SPDLT ,TB2C(9,4) ,XV (12) ESTM0066
2 ,OBS(16,4) ,STA(10) ,TT2BO (9) ESTM0067
EQUIVALENCE (CDAT(293),IDATA ) ,(CDAT(159),TAU ) ESTM0068
1 ,(CDAT( 5),OBS ) ,(CDAT(161),TB2C ) ESTM0069
2 ,(CDAT(135),OMEGA ) ,(CDAT(233),TT2BO ) ESTM0070
3 ,(CDAT( 1),BIAS ) ,(CDAT(137),SPDLT ) ,(CDAT(251),TT2BT ) ESTM0071
4 ,(CDAT(133),FTR ) ,(CDAT(139),STA ) ,(CDAT(269),XV ) ESTM0072
ESTM0073
ESTM0074

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EQUIVALENC (IDATA( 2),MODE ) ,(IDATA( 5),NALIGN) ESTM0075
COMMON SAVE(90),STT(45,2) ESTM0076
DOUBLE PRECISION STT ESTM0077
C
DOUBLE PRECISION D(36) ,ETIMR ,GHA ,PEH(30) ,TNPT1 ESTM0079
1 ,DELT ,ETIMT ,FLT(8) ,PH (36) ,UTIMR ESTM0080
2 ,DELT ,ETIMV ,HI(30) ,SLRAN2 ,W (7) ESTM0081
3 ,EP ESTM0082
DIMENSION DATA(4) ,IQL (4) ,LBTYP(4) ,PRCD (5) ESTM0083
1 ,FIL(12) ,JDATA (7) ,NM (4) ,SYM(4,4) ESTM0084
2 ,FMT (8) ,LBM(12,4) ,NN (8) ESTM0085
LOGICAL L ,LIN ,LOUTL ESTM0086
EQUIVALENC (D(1),PEH) , (W(1),TIME) , (QUAL,IQUAL) ESTM0087
DATA EP /3.D-4/ ESTM0088
DATA FLT /1.D0,2.D0,3.D0,4.D0,5.D0,6.D0,7.D0,8.D0/ ESTM0089
DATA JDATA/ 1, 0, 1, 1, 0, 1, 1/ ESTM0090
DATA NM /-3,1,3,-1/ ESTM0091
DATA NN / 0, 6,20,24,26,50,74,84/ ESTM0092
DATA FIL /6H 9,6H 1X,E1,6H14X,E1,6H27X,E1,6H40X,E1 ESTM0093
1 ,6HF7.1,2,6H3.5,40,6H3.5,27,6H3.5,14,6H3.5, 1 ESTM0094
2 ,6H0 ,6H / ESTM0095
DATA FMT /6H(A1, ,6HF7.1,2,6HX,A3,E,6H17.8, ,6H X,E1 ESTM0096
1 ,6H3.5, ,6HX,3E12,6H.4) / ESTM0097
DATA LBM /6HAZIMUT,6HELEVAT,6HRANGE ,6H ,6HH ,6HION ESTM0098
1 ,6H ,6H ,6HRAD ,6HRAD ,6HKM ,6H ESTM0099
2 ,6HX ,6HY ,6HRANGE ,6HDOPPLE,6H ,6H ESTM0100
3 ,6H ,6HR ,6HRAD ,6HRAD ,6HSEC ,6HCYCLESESTM0101
3 ,6HX ,6HY ,6HRANGE ,6HDOPPLE,6H ,6H ESTM0102
2 ,6H ,6HR ,6HRAD ,6HRAD ,6HKM ,6HCYCLESESTM0103
4 ,6HHOUR A,6HDECLIN,6H ,6HDOPPLE,6HNGLE ,6HATION ESTM0104
4 ,6H ,6HR ,6HRAD ,6HRAD ,6H ,6HCYCLESESTM0105
5 / ESTM0106
DATA LRTYP/6HC-BAND,6HG-RR ,6HS-BAND,6HDSIF / ESTM0107
DATA PRCD /6H, PROC,6HEED TO,6H THE N,6HEXT PR,6HOCESS / ESTM0108
DATA START ,STOP /6HOSTART,6H STOP / ESTM0109
DATA SYM /3HAZM,3HELV,3HRGE,3H ,3HX ,3HY ,3HRGE,3HDOP ESTM0110
1 ,3HX ,3HY ,3HRGE,3HDOP ,3HHRA,3HDEC,3H ,3HDOP / ESTM0111
601 FORMAT(/23HORECEIVING STATION ,A6/23H TRANSMITTING STATION ,ESTM0112
1A6/10H DATA TYPE,13X,A6,2H (,3(2A6,2H, ),2A6/32X,4(A6,8X),1H)) ESTM0113
602 FORMAT(15H MEAS VARIANCES,13X,4E14.4) ESTM0114
603 FORMAT(/41HOTIME TAGS ARE REFERRED TO FIRST ONTIME =,D24.16,23H SEESTM0115
1CONDS (ST) FROM 1950) ESTM0116
604 FORMAT(A6,14H PROCESSING AT,F12.2,13H SECONDS (ST)) ESTM0117
605 FORMAT(35H STOP PROCESSING AT END OF DATA ARC) ESTM0118
606 FORMAT(6HOSKIP ,I3,31H DATA PTS BETWEEN PROCESSED PTS) ESTM0119
609 FORMAT(/26HO TIME OBS MEASURED,28X,9HRESIDUALS,24X,5HHZHAT,ESTM0120
17X,5HHZBAR,7X,4HHPHT) ESTM0121
612 FORMAT(31HO** ALL MEASUREMENTS SUPPRESSED,5A6) ESTM0122
613 FORMAT(21HONNEW DATA RECORD READ) ESTM0123
641 FORMAT(/30HOLINEARITY TEST FAILED, ITER =,I2) ESTM0124
642 FORMAT(/30HOLINEARITY TEST PASSED, ITER =,I2) ESTM0125
C ESTM0126
C**** INITIALIZATION ESTM0127
C ESTM0128
1 IF (LTRAJ) GO TO 60 ESTM0129
C ESTM0130
C HERE FOR INITIAL ENTRY FOR EACH PROCESS ESTM0131
C LOAD A PRIORI COVARIANCE MATRIX AND STATE INDICES ESTM0132
C ESTM0133
10 CONTINUE ESTM0134
IF (NPREST.EQ.0) GO TO 11 ESTM0135
READ (12) DEST ESTM0136
WRITE (9) DEST ESTM0137
NPREST = 0 ESTM0138
GO TO 12 ESTM0139
11 READ (9) DEST ESTM0140
12 BACKSPACE 9 ESTM0141
ISTAL = 1 ESTM0142
LSTA = .TRUE. ESTM0143
J = 100+INDSTA ESTM0144
DO 13 I=1,53,2 ESTM0145
IF (KLOCN(I).EQ.0) GO TO 15 ESTM0146
IF (KLOCN(I)-J) 13,14,15 ESTM0147
13 ISTAL = ISTAL+KLOCN(I+1) ESTM0148
GO TO 15 ESTM0149

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14	ISTA2 = ISTA1+KLNOCN(I+1)-1	ESTM0150
	LSTA = .FALSE.	ESTM0151
15	CONTINUE	ESTM0152
	DO 16 I=1,4	ESTM0153
	EPSK(I) = EPSDC*SE(I+10,INDSTA)	ESTM0154
16	Q2(I) = SE(I+10,INDSTA)**2	ESTM0155
C		ESTM0156
C	SET UP COUNTERS AND REFERENCES	ESTM0157
20	CONTINUE	ESTM0158
	III = NPSKIP	ESTM0159
	NPSKIP = NPSKIP+1	ESTM0160
	TFIRST = ONTIME	ESTM0161
	STPTIM = STIMR(2)-ONTIME	ESTM0162
	DO 21 I=1,7	ESTM0163
21	IDATA(I) = JCATA(I)	ESTM0164
	ASSIGN 801 TO N122	ESTM0165
	ASSIGN 22 TO N129	ESTM0166
	GO TO 110	ESTM0167
22	ASSIGN 800 TO N122	ESTM0168
	ASSIGN 130 TO N129	ESTM0169
	ASSIGN 23 TO N130	ESTM0170
23	CONTINUE	ESTM0171
	STP = STIMR(1)-ONTIME	ESTM0172
	DO 24 NPT=1,NPTS	ESTM0173
24	IF (STP.LE.BUFDAT(NPT,2)) GO TO 25	ESTM0174
	NPT = NPTS+1	ESTM0175
	GO TO 100	ESTM0176
25	ASSIGN 131 TO N130	ESTM0177
	NPT1 = NPT-NPSKIP	ESTM0178
	NPT2 = NPT1	ESTM0179
	NDEL1 = NPEND+2	ESTM0180
	NINT = NPEND+1	ESTM0181
	NHU = NEMPS+6	ESTM0182
	YMOD(1) = CON(3)	ESTM0183
	YMOD(2) = CON(3)	ESTM0184
	YMOD(4) = YTEST	ESTM0185
	OBIAS(4) = 0.DO	ESTM0186
C		ESTM0187
C	WRITE HEADING DATA	ESTM0188
30	CONTINUE	ESTM0189
	WRITE (6,601) NRSTA ,NTSTA ,LBTYP(MTYPE)	ESTM0190
	1 , (LABEL(I),LABEL(I+4),I=1,4) , (LABEL(I),I=9,12)	ESTM0191
	WRITE (6,602) Q2	ESTM0192
	WRITE (6,603) TFIRST	ESTM0193
	CALL DATOUP (TFIRST,D,0)	ESTM0194
	WRITE (6,604) START ,BUFDAT(NPT,2)	ESTM0195
	GO TO (31,32) ,IFTSTP	ESTM0196
31	WRITE (6,604) STCP ,STPTIM	ESTM0197
	ASSIGN 201 TO N200	ESTM0198
	GO TO 33	ESTM0199
32	WRITE (6,605)	ESTM0200
	ASSIGN 202 TO N200	ESTM0201
33	CONTINUE	ESTM0202
	IF (III.NE.0) WRITE (6,606) III	ESTM0203
C		ESTM0204
C	SET SWITCHES	ESTM0205
40	CONTINUE	ESTM0206
	IF (IFOUTL.NE.0) GO TO 41	ESTM0207
	ASSIGN 203 TO N203	ESTM0208
	ASSIGN 349 TO N302	ESTM0209
	KOUTL = .TRUE.	ESTM0210
	GO TO 42	ESTM0211
41	ASSIGN 204 TO N203	ESTM0212
	ASSIGN 303 TO N302	ESTM0213
	KOUTL = .FALSE.	ESTM0214
42	IF (IFBADD.NE.0) GO TO 43	ESTM0215
	ASSIGN 100 TO N204	ESTM0216
	ASSIGN 301 TO N300	ESTM0217
	GO TO 44	ESTM0218
43	ASSIGN 205 TO N204	ESTM0219
	ASSIGN 302 TO N300	ESTM0220
44	CONTINUE	ESTM0221
C		ESTM0222
C	LOAD FIXED DATA	ESTM0223
50	CONTINUE	ESTM0224

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DO 51 I=1,9
TT2BO(I) = SN(I+4,INDSTA)
51 TT2BT(I) = SN(I+4,ITRSTA)
GO TO 70
C
C   HERE FOR REPEATED ENTRIES
60 CONTINUE
READ (9) DEST
BACKSPACE 9
C
C   LOAD STATE-DEPENDENT DATA
70 CONTINUE
DO 71 I=1,KOUNTN
71 HI(I) = 0.DO
CALL DMVTRN (TT2BO,SE(1,INDSTA),STA ,1,1)
CALL DMVTRN (TT2BT,SE(1,ITRSTA),STA(6),1,1)
DO 72 I=1,3
STA(I) = SN(I+1,INDSTA)+STA(I)
72 STA(I+5) = SN(I+1,ITRSTA)+STA(I+5)
BIAS(1) = DATIND(15)+SE(10,INDSTA)/TAU
SPDLT = CON(8)+SE(6,INDSTA)
DO 73 I=1,3
73 OBIAS(I) = SE(I+6,INDSTA)
C
C   CONTINUE PROCESSING DATA
90 CONTINUE
LTRAJ = .FALSE.
NPT = NP11
TNPT1 = TBF
DELTP = VBF(NDELTP,1)
WRITE (6,609)
C
C**** DATA INPUT CONTRCL
C
100 NPT = NPT+NPSKIP
IF (NPT.LE.NPTS) GO TO 200
NPT = NPT-NPTS
C
C   LOCATE NEXT DATA RECORD
ASSIGN 102 TO N101
101 READ (10) INDDAT
IF (NEOT.EQ.0) GO TO N101, (102,103)
GO TO 800
102 IF (KONT.NE.0) GO TO 130
NEW STATION PASS
ASSIGN 103 TO N101
NPT = 1
GO TO (103,800) ,IFTSTP
103 IF (STIMR(2).LT.ONTIME) GO TO 800
IF (NPRSTA.EQ.NRSTA) GO TO 110
READ (10) SKIP
GO TO 101
C
C   SET DATA TAPE DEPENDENT VARIABLES
110 CONTINUE
TREF = ONTIME-TFIRST
NALIGN = INDDAT( 9)
MODE = INDDAT(10)
TAU = DATIND(10)
FTR = DATIND(11)
STA(4) = DATIND(12)
STA(5) = DATIND(13)
YMOD(3) = DATIND(14)
BIAS(1) = DATIND(15)+SE(10,INDSTA)/TAU
BIAS(2) = DATIND(16)
GO TO (111,112,113,114) ,MTYPE
111 ASSIGN 291 TO N290
MODE = 0
GO TO 120
112 ASSIGN 292 TO N290
GO TO 120
113 ASSIGN 293 TO N290
GO TO 120
114 ASSIGN 294 TO N290
C
ESTM0225
ESTM0226
ESTM0227
ESTM0228
ESTM0229
ESTM0230
ESTM0231
ESTM0232
ESTM0233
ESTM0234
ESTM0235
ESTM0236
ESTM0237
ESTM0238
ESTM0239
ESTM0240
ESTM0241
ESTM0242
ESTM0243
ESTM0244
ESTM0245
ESTM0246
ESTM0247
ESTM0248
ESTM0249
ESTM0250
ESTM0251
ESTM0252
ESTM0253
ESTM0254
ESTM0255
ESTM0256
ESTM0257
ESTM0258
ESTM0259
ESTM0260
ESTM0261
ESTM0262
ESTM0263
ESTM0264
ESTM0265
ESTM0266
ESTM0267
ESTM0268
ESTM0269
ESTM0270
ESTM0271
ESTM0272
ESTM0273
ESTM0274
ESTM0275
ESTM0276
ESTM0277
ESTM0278
ESTM0279
ESTM0280
ESTM0281
ESTM0282
ESTM0283
ESTM0284
ESTM0285
ESTM0286
ESTM0287
ESTM0288
ESTM0289
ESTM0290
ESTM0291
ESTM0292
ESTM0293
ESTM0294
ESTM0295
ESTM0296
ESTM0297
ESTM0298
ESTM0299

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C	SET DATA TAPE DEPENDENT SWITCHES	ESTM0300
120	CONTINUE	ESTM0301
	L = .FALSE.	ESTM0302
	DO 122 I=1,4	ESTM0303
	LSKIP(I) = .TRUE.	ESTM0304
	DO 121 J=I,12,4	ESTM0305
121	LABEL(J) = LBM(J,MTYPE)	ESTM0306
	SYMBOL(I) = SYM(I,MTYPE)	ESTM0307
	IF (MSKIP(I,MTYPE)+IFSKIP(I).GT.0) GO TO 122	ESTM0308
	LSKIP(I) = .FALSE.	ESTM0309
	L = .TRUE.	ESTM0310
122	CONTINUE	ESTM0311
	IF (L) GO TO 123	ESTM0312
	WRITE (6,612) PRCD	ESTM0313
	GO TO N122, (800,801)	ESTM0314
123	STA(9) = STA(4)	ESTM0315
	STA(10) = STA(5)	ESTM0316
129	GO TO N129, (22,130)	ESTM0317
C		ESTM0318
C	READ NEXT DATA RECORD	ESTM0319
130	READ (10) BUFDAT	ESTM0320
	GO TO N130, (23,131)	ESTM0321
C		ESTM0322
C	STORE NEW A PRIORI ESTIMATE WITH EXISTING NOMINAL	ESTM0323
131	CONTINUE	ESTM0324
	NPT1 = NPT-NPSKIP	ESTM0325
	NPT2 = NPT1	ESTM0326
132	WRITE (9) DEST	ESTM0327
	BACKSPACE 9	ESTM0328
	DO 133 I=1,KGUNTN	ESTM0329
	ZBAR(I) = ZBAR(I)-ZHAT(I)	ESTM0330
	ZTOT(I) = ZTOT(I)+ZHAT(I)	ESTM0331
133	ZHAT(I) = 0.00	ESTM0332
	IF (ITER.NE.0) GO TO 425	ESTM0333
	WRITE (6,613)	ESTM0334
	CALL ESTCUT	ESTM0335
C		ESTM0336
C****	COMPUTE OBSERVABLES AND PARTIALS	ESTM0337
C		ESTM0338
C	SET UP WORKING ARRAYS	ESTM0339
200	CONTINUE	ESTM0340
	TIME = BUFDAT(NPT,2)+TREF	ESTM0341
	GO TO N200, (201,202)	ESTM0342
201	IF (STPTIM.LT.TIME) GO TO 801	ESTM0343
202	QUAL = BUFDAT(NPT,1)	ESTM0344
	LCUTL = KOUTL	ESTM0345
	DO 204 I=1,4	ESTM0346
	DATA(I) = BUFDAT(NPT,I+2)	ESTM0347
	GO TO N203, (203,204)	ESTM0348
203	IF (DATA(I).GE.0.) LOUFL = .FALSE.	ESTM0349
204	IQL(I) = 0	ESTM0350
	IF (LCUTL) GO TO 100	ESTM0351
	IF (IQUAL.EQ.0) GO TO 206	ESTM0352
	IF (IQUAL.EQ.7) GO TO N204, (100,205)	ESTM0353
205	IQL(4) = IQUAL/4	ESTM0354
	IQL(1) = IQUAL-4*IQL(4)	ESTM0355
	IQL(3) = IQL(1)/2	ESTM0356
	IQL(2) = IQL(1)-2*IQL(3)	ESTM0357
	IQL(1) = IQL(2)	ESTM0358
206	CONTINUE	ESTM0359
	IF (KTAU.NE.0) GO TO 207	ESTM0360
	TAU = DATA(4)	ESTM0361
	DATA(4) = DATIND(10)	ESTM0362
	BIAS(1) = DATIND(15)+SE(10,INDSTA)/TAU	ESTM0363
207	CONTINUE	ESTM0364
C		ESTM0365
C	COMPUTE VEHICLE TIME AND STATE	ESTM0366
C		ESTM0367
C	RECEPTION TIME	ESTM0368
210	CONTINUE	ESTM0369
	STIMR = TFIRST+TIME	ESTM0370
	UTIMR = STIMR+SE(4,INDSTA)+SE(5,INDSTA)*STIMR	ESTM0371
	ETIMR = UTIMR+DELDAN(1)+DELDAN(2)*UTIMR	ESTM0372
	III = 1	ESTM0373
	D(2) = (ETIMR-TBF)/HRF	ESTM0374

C	T**N/N-FACTORIAL	ESTM0375
C	220 D(1) = FLT(1)	ESTM0376
	DO 221 I=2,8	ESTM0377
	221 D(I+1) = D(2)*D(I)/FLT(I)	ESTM0378
	GO TO (230,251,262) ,III	ESTM0379
C	COMPUTE DOWN-LEG DELAY TIME	ESTM0380
C	230 DELT = 0.00	ESTM0381
	DO 231 I=1,8	ESTM0382
	231 DELT = DELT+D(I)*VBF(NDELT,I)	ESTM0383
	ETIMV = ETIMR-DELT	ESTM0384
	ETIMT = ETIMV-DELT	ESTM0385
	IF (ETIMV.LE.TFF) GO TO 250	ESTM0386
C	SET ICS FOR NEW INTERPOLATION TABLES	ESTM0387
C	240 CONTINUE	ESTM0388
	NPT1 = NPT-NPSKIP	ESTM0389
	NPT2 = NPT1	ESTM0390
	WRITE (9) DEST	ESTM0391
	BACKSPACE 9	ESTM0392
	TBF = ETIMV	ESTM0393
	VBF(NDELT,1) = DELT	ESTM0394
	242 LTRAJ = .TRUE.	ESTM0395
	N = 0	ESTM0396
	DO 244 I=1,53,2	ESTM0397
	J = KLOCN(I)	ESTM0398
	K = KLOCN(I+1)	ESTM0399
	IF (J.EQ.0) GO TO 245	ESTM0400
	JJ = 0	ESTM0401
	IF (J.LT.100) GO TO 243	ESTM0402
	JJ = J-100	ESTM0403
	J = 8	ESTM0404
	243 DO 244 III=1,K	ESTM0405
	N = N+1	ESTM0406
	M = ILOCN(N)+NN(J)+14*JJ	ESTM0407
	SPCDAN(M) = SPCDAN(M)-ZBAR(N)+ZHAT(N)	ESTM0408
	ZBAR(N) = ZHAT(N)	ESTM0409
	IF (ITER.NE.0) GO TO 244	ESTM0410
	ZTOT(N) = ZTOT(N)+ZHAT(N)	ESTM0411
	ZBAR(N) = 0.00	ESTM0412
	244 ZHAT(N) = 0.00	ESTM0413
	245 CONTINUE	ESTM0414
	GO TO 999	ESTM0415
C	INTERPOLATE FOR VEHICLE STATE	ESTM0416
C	250 III = 2	ESTM0417
	D(2) = (ETIMV-TBF)/HBF	ESTM0418
	GO TO 220	ESTM0419
	251 CONTINUE	ESTM0420
	CALL DMPY (VBF,D,STT,NINT,8,2,45,1,45,0)	ESTM0421
	DO 252 I=1,NINT	ESTM0422
	252 STT(I,2) = RBF(I)+HBF*STT(I,2)	ESTM0423
	CALL DEQTR (ETIMV,D(10))	ESTM0424
	D(2) = STT(NPEND+1,1)+D(12)	ESTM0425
	D(3) = STT(NPEND,1)	ESTM0426
	D(4) = D(12)	ESTM0427
	D(5) = D(3)*DCOS(D(2))	ESTM0428
	D(6) = ETIMV-DELDAN(1)-DELDAN(2)*ETIMV	ESTM0429
	CALL DEHA (D(6),D(5),D(1),OMEGA)	ESTM0430
	GHA = D(1)	ESTM0431
	CALL DGTRN (D(16),NM,D,4)	ESTM0432
	CALL DGTSN (D(25),0,D(11),D(10),D(13))	ESTM0433
	CALL DMVTRN (D(25),D(16),D,2,3)	ESTM0434
	DO 254 I=1,9	ESTM0435
	254 D(I+9) = 0.00	ESTM0436
	DO 255 I=1,9,4	ESTM0437
	255 D(I+9) = 1.00	ESTM0438
	D(19) = ETIMV-ETIMR	ESTM0439
	D(21) = ETIMV-ETIMT	ESTM0440
	D(20) = D(19)+TAU	ESTM0441
	D(22) = D(21)+TAU	ESTM0442
	DO 256 I=1,4	ESTM0443
	D(10) = DCOS(OMEGA*D(I+18))	ESTM0444
	D(13) = DSIN(OMEGA*D(I+18))	ESTM0445
		ESTM0446
		ESTM0447
		ESTM0448
		ESTM0449

	D(11) = -D(13)	ESTM0450
	D(14) = D(10)	ESTM0451
256	CALL DMVTRN (D,C(10),TB2C(1,1),1,3)	ESTM0452
C		ESTM0453
C	LOAD SPACECRAFT STATE	ESTM0454
260	CONTINUE	ESTM0455
	DO 261 I=1,3	ESTM0456
	XV(I) = STT(I,2)	ESTM0457
261	XV(I+3) = STT(I,1)	ESTM0458
	D(2) = (ETIMV-TBF-TAU)/HBF	ESTM0459
	III = 3	ESTM0460
	GO TC 220	ESTM0461
262	CONTINUE	ESTM0462
	CALL DMPY (VBF,D(2),XV(7),3,8,2,45,-1,3,0)	ESTM0463
	DO 263 I=1,3	ESTM0464
263	XV(I+6) = XV(I+6)*HBF+RHF(I)	ESTM0465
C		ESTM0466
C	COMPUTE OBSERVABLES AND PARTIALS	ESTM0467
290	CONTINUE	ESTM0468
	GO TO N290, (291,292,293,294)	ESTM0469
C	C-BAND	ESTM0470
291	CALL CBDATP (SLRAN2)	ESTM0471
	GO TO 300	ESTM0472
C	GDDARD	ESTM0473
292	CALL GRDATP (SLRAN2)	ESTM0474
	GO TC 300	ESTM0475
C	S-BAND	ESTM0476
293	CALL SBDATP (SLRAN2)	ESTM0477
	GO TC 300	ESTM0478
C	DSIF	ESTM0479
294	CALL DSDATP (GHA,SLRAN2)	ESTM0480
	GO TC 300	ESTM0481
C		ESTM0482
C****	COMPUTE RESIDUALS AND DIFFERENTIAL CORRECTIONS	ESTM0483
C		ESTM0484
C	RESIDUALS FROM THE NOMINAL	ESTM0485
300	CONTINUE	ESTM0486
	III = 1	ESTM0487
	FMT(2) = FIL(6)	ESTM0488
	DO 349 I=1,4	ESTM0489
	IF (LSKIP(I)) GO TO 349	ESTM0490
	IF (DATA(I).EQ.YTEST) GO TO 349	ESTM0491
	GO TO N300, (301,302)	ESTM0492
301	IF (IQL(I).GT.0) GO TO 349	ESTM0493
302	W(3) = DATA(I)	ESTM0494
	IF (DATA(I).GE.0.) GO TO 304	ESTM0495
	GO TC N302, (303,349)	ESTM0496
303	W(3) = -W(3)	ESTM0497
304	W(4) = W(3)-CRS(1,I)-GRIAS(I)	ESTM0498
	W(2) = SYMBOL(I)	ESTM0499
C		ESTM0500
C	COMPUTE PARTIALS WRT STATE	ESTM0501
310	CONTINUE	ESTM0502
	IF (NHU.EQ.0) GO TO 320	ESTM0503
	CALL DMPY (OBS(2,I),STT(4,2),PH,1,3,NHU,1,3,1,0)	ESTM0504
	CALL DMPY (CRS(2,I),STT(4,1),PEH,1,3,NHU,1,3,1,0)	ESTM0505
	J = 1	ESTM0506
	N = NHU	ESTM0507
	IF (KLOCN(1).EQ.1) GO TO 312	ESTM0508
	J = 7	ESTM0509
	N = NEMPS	ESTM0510
312	CONTINUE	ESTM0511
	DO 313 K=1,N	ESTM0512
	H1(K) = PH(J)+PEH(J)	ESTM0513
313	J = J+1	ESTM0514
C		ESTM0515
C	COMPUTE PARTIALS WRT STATION ERRORS	ESTM0516
320	CONTINUE	ESTM0517
	IF (LSTA) GO TO 330	ESTM0518
	DO 329 M=ISTA1,ISTA2	ESTM0519
	K = ILOCN(M)	ESTM0520
	GO TO (321,321,321,322,323,324,325,325,325,325) ,K	ESTM0521
321	H1(M) = CBS(K+10,I)	ESTM0522
	GO TO 329	ESTM0523
322	H1(M) = CRS(9,I)	ESTM0524

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GO TO 329
323 H1(M) = OBS(9,I)*STIMR
GO TO 329
324 H1(M) = OBS(10,I)
GO TO 329
325 IF (K.NE.I+6) GO TO 329
H1(M) = OBS(8,I)
329 CONTINUE
C
C RESIDUALS FROM THE CURRENT ESTIMATE
330 W(5) = 0.00
W(6) = 0.00
W(7) = 0.00
DO 332 M=1,KOUNTN
W(5) = W(5)+H1(M)*ZHAT(M)
W(6) = W(6)+H1(M)*ZBAR(M)
PH(M) = 0.00
DO 331 K=1,KOUNTN
331 PH(M) = PH(M)+PNEW(M,K)*H1(K)
PEH(M) = PH(M)+EPSK(I)*H1(M)
332 W(7) = W(7)+PH(M)*H1(M)
D(33) = W(7)+Q2(I)
W(4) = W(4)-W(5)+W(6)
STP = W(4)/YMOD(I)+0.500
IF (STP.LT.0.) STP = STP-1.
STP = AINT(STP)
W(4) = W(4)-STP*YMOD(I)
C
C COMPUTE DIFFERENTIAL CORRECTION
340 CONTINUE
D(32) = W(4)/D(33)
D(31) = EPSK(I)**2
DO 347 J=1,KOUNTN
IF (ITRETN(J).EQ.0) GO TO 341
ASSIGN 344 TO N342
ASSIGN 343 TO N343
ZHAT(J) = ZHAT(J)+D(32)*PEH(J)
GO TO 342
341 ASSIGN 346 TO N342
ASSIGN 344 TO N343
342 DO 346 K=J,KOUNTN
IF (ITRETN(K).EQ.0) GO TO N342, (344,346)
GO TO N343, (343,344)
343 PNEW(J,K) = PNEW(J,K)-(PH(J)*PH(K)-D(31)*H1(J)*H1(K))/D(33)
GO TO 345
344 PNEW(J,K) = PNEW(J,K)-PH(J)*PEH(K)/D(33)
345 PNEW(K,J) = PNEW(J,K)
346 CONTINUE
347 CONTINUE
FMT(5) = FIL(I+1)
FMT(6) = FIL(I+6)
WRITE (6,FMT) FIL(III+10),(W(K),K=III,7)
III = 2
FMT(2) = FIL(1)
349 CONTINUE
C
C**** LINEARITY CONTROL
C
C LINEARITY TEST
400 CONTINUE
IF (NPT.LT.NPT2) GO TO 100
DO 401 I=1,KOUNTN
401 D(I) = ZBAR(I)-ZHAT(I)
CALL DMPY (STT(4,2),D,PH,3,NHU,1,3,1,1,0)
D(4) = DDOT(PH,PH)
LIN = .TRUE.
III = 1
IF (NPT.NE.NPT2) GO TO 402
III = 2
IF (D(4).GT.DPREV) GO TO 998
402 IF (D(4).GT.SLRAN2*EP) LIN = .FALSE.
IF (D(4).GT.CBRAN2*EP) LIN = .FALSE.
IF (LIN) GO TO (100,420) ,III
C
C ITERATION CONTROL

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ESTM0525
ESTM0526
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ESTM0575
ESTM0576
ESTM0577
ESTM0578
ESTM0579
ESTM0580
ESTM0581
ESTM0582
ESTM0583
ESTM0584
ESTM0585
ESTM0586
ESTM0587
ESTM0588
ESTM0589
ESTM0590
ESTM0591
ESTM0592
ESTM0593
ESTM0594
ESTM0595
ESTM0596
ESTM0597
ESTM0598
ESTM0599

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C	RESET A PRIORI AND NOMINAL FOR NEW ITERATION	ESTM0600
410	DPREV = D(4)	ESTM0601
	TBF = INPT1	ESTM0602
	VBF(NDELT,1) = DELTP	ESTM0603
	READ (9) DEST	ESTM0604
	BACKSPACE 9	ESTM0605
	WRITE (6,641) ITER	ESTM0606
	ITER = ITER+1	ESTM0607
	IF (ITER.GT.6) GC TO 998	ESTM0608
	NPT2 = NPT	ESTM0609
	GO TO 242	ESTM0610
C		ESTM0611
C	ACCEPT ESTIMATE AS NEW A PRIORI	ESTM0612
420	NPT1 = NPT2	ESTM0613
	TNPT1 = ETIMV	ESTM0614
	DELTP = DELT	ESTM0615
	GO TO 132	ESTM0616
425	WRITE (6,642) ITER	ESTM0617
	ITER = 0	ESTM0618
	GO TO 100	ESTM0619
C		ESTM0620
C****	POSTLOGUE	ESTM0621
C		ESTM0622
800	BACKSPACE 10	ESTM0623
801	NTP = 4	ESTM0624
	STIMR(2) = STIMR(1)*1.000000100	ESTM0625
	GO TO 240	ESTM0626
C		ESTM0627
C	EXIT	ESTM0628
998	IERR = 18	ESTM0629
999	RETURN	ESTM0630
	END	

Subroutine: ESTOUT

Purpose: To write the state estimate and covariance matrix on
the system output tape and on the estimate tape.

Calling Sequence: CALL ESTOUT

Common storages used: // 20 cells, /DCPCØM/, /DCRCØM/, /ESTCØM/,
/ES1CØM/, /SBFCØM/

Subroutines required: CØVØUT, DATØUP

ESTØUT-1

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$IBFTC MC1330 XR3,M94,NOCD,LIST
SUBROUTINE ESTOUT
C OUTPUTS ESTIMATE AND COVARIANCE MATRIX
C
COMMON /DCPCOM/CDPC(900)
EQUIVALENCE (CDCP(777),ICB ),(CDCP(117),NESEND)
1 ,(CDCP(678),NCVOUT) ,(CDCP(742),NPRSTO)
C
COMMON /ESTCOM/CEST(804)
DOUBLE PRECISION ETIMVA,SPCDAN(6)
EQUIVALENCE (CEST( 27),KOUNTN) ,(CEST( 1),NESPOS)
1 ,(CEST( 29),ETIMVA) ,(CEST( 2),NBY ) ,(CEST( 49),SPCDAN)
C
COMMON /ESICOM/PNEW(30,30),STNAMN(30),TRAKER(30)
1 ,ITRETN (30),ILOCN (30),KLOCN (54)
2 ,NSN (12,20)
DOUBLE PRECISION PNEW
DIMENSION DEST(2214)
EQUIVALENCE (PNEW,DEST)
C
COMMON /SRFCOM/CSBF(12),RBF(45),VBF(45,8)
DOUBLE PRECISION RBF,VBF,TBF
EQUIVALENCE (CSBF(2),NTP),(CSBF(6),ITER),(CSBF(9),TRF)
C
COMMON /DCRCOM/ZBAR(30),ZHAT(30),ZTOT(30),DCSAVE(74)
DOUBLE PRECISION ZBAR ,ZHAT ,ZTOT
C
COMMON D(30)
DOUBLE PRECISION D
C
DIMENSION NST(8),STN(6)
DATA NST/0,6,20,24,26,50,74,84/
DATA STN/36HX Y Z XD ZD /
601 FORMAT(1H /18HOA PRIORI ESTIMATE/10H ETIMVA = ,D23.16)
602 FORMAT(14HOITERATION NR ,12/10H ETIMVA = ,D23.16)
603 FORMAT(29HOESTIMATE NOT STORED ON TAPE /10H ETIMVA = ,D23.16)
604 FORMAT(35HOCUMULATIVE DIFFERENTIAL CORRECTION)
605 FORMAT(3(3X,A6,D24.16))
606 FORMAT(15HSTATE ESTIMATE)
607 FORMAT(1H /13HOESTIMATE NR ,13,15H STORED ON TAPE/10H ETIMVA = ,D23.16)
C
C IDENTIFY OUTPUT REQUIREMENTS
1 GO TO (10,20,25,30) ,NTP
C
C PROPAGATION END-POINT
10 IF (NCVCUT.LE.0) GO TO 999
GO TO 51
C
C ESTABLISH ANCHOR POINT
20 NBY = ICB+100
ETIMVA = TBF
DO 21 I=1,3
SPCDAN(I) = RBF(I)
21 SPCDAN(I+3) = VBF(I,1)
WRITE (6,601) ETIMVA
CALL DATCUP (ETIMVA,D(6),0)
GO TO 37
25 WRITE (6,602) ITER,ETIMVA
GO TO 35
C
C STORE ESTIMATE ON TAPE
30 IF (NPRSTO.NE.0) GO TO 31
WRITE (6,603) ETIMVA
GO TO 34
31 NN = NESEND-NESPOS
IF (NN.EQ.0) GO TO 33
DO 32 I=1,NN
READ (12) SKIP
32 READ (12) SKIP
33 NESPOS = NESEND+1
NESEND = NESPOS
WRITE (12) CEST
WRITE (12) DEST
WRITE (6,607) NESPOS,ETIMVA
EST00001
EST00002
EST00003
EST00004
EST00005
EST00006
EST00007
EST00008
EST00009
EST00010
EST00011
EST00012
EST00013
EST00014
EST00015
EST00016
EST00017
EST00018
EST00019
EST00020
EST00021
EST00022
EST00023
EST00024
EST00025
EST00026
EST00027
EST00028
EST00029
EST00030
EST00031
EST00032
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EST00068
EST00069
EST00070
EST00071
EST00072
EST00073
EST00074

```

34 CALL DATCUP (ETIMVA,D(6),0)	EST00075
35 WRITE (6,604)	EST00076
DO 36 I=1,KOUNTN	EST00077
36 D(I) = ZTOT(I)+ZHAT(I)	EST00078
WRITE (6,605) (STNAMN(I),D(I),I=1,KOUNTN)	EST00079
IF (NTP.EQ.0) GO TO 999	EST00080
WRITE (6,606)	EST00081
37 WRITE (6,605) (STN(I),SPCDAN(I),I=1,6)	EST00082
NN = 1	EST00083
IF (KLOCN(1).EQ.1) NN=KLOCN(2)+1	EST00084
IF (KOUNTN.LT.NN) GO TO 50	EST00085
MM = NN	EST00086
DO 41 I=1,53,2	EST00087
K = KLOCN(I)	EST00088
L = KLOCN(I+1)	EST00089
IF (K-1) 42,41,38	EST00090
38 KK = 0	EST00091
IF (K.LT.100) GO TO 39	EST00092
KK = K-100	EST00093
K = 8	EST00094
39 JJ = NST(K)+14*KK	EST00095
DO 40 J=1,L	EST00096
M = JJ+ILOCN(MM)	EST00097
D(MM) = SPCDAN(M)	EST00098
40 MM = MM+1	EST00099
41 CONTINUE	EST00100
42 CONTINUE	EST00101
WRITE (6,605) (STNAMN(I),D(I),I=NN,KOUNTN)	EST00102
C	EST00103
C OUTPUT COVARIANCE MATRIX	EST00104
50 WRITE (9) DEST	EST00105
BACKSPACE 9	EST00106
IF (NCVOUT.LE.0) GO TO 999	EST00107
51 DO 52 I=1,3	EST00108
D(I) = SPCDAN(I) +RBF(I+42)	EST00109
52 D(I+3) = SPCDAN(I+3)+VBF(I+42,1)	EST00110
CALL COVOUT (KOUNTN,D,NCVOUT)	EST00111
C	EST00112
999 RETURN	EST00113
END	

Subroutine: EXINST

Purpose: To read interpolation coefficients from a binary tape (as written by XØUT) and to reconstruct an N-dimensional vector X (and its derivative, if any) at time T.

Calling Sequence: CALL EXINST(T,M,N,X,XD,W,C,NB,ITRIG,KØUNT)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	T	d			Time, or any independent variable.
I/Ø	M				Vector number and indicator.
I/Ø	N				Vector dimension and indicator.
Ø	X	(N)			Vector array.
Ø	XD	(N)			Derivative array.
Ø	W	6K [*]			Internal storage.
Ø	C	As required			Critical information array.
I	NB				Binary tape number.
I	ITRIG				Trigger for KØUNT. KØUNT is altered only when ITRIG = 1.
I/Ø	KØUNT				Logical record counter. Reading forward increments KØUNT, backspacing decrements KØUNT.

Common storages used: None

Subroutines required: BACK

*K is the total of all vector dimensions (not including derivatives).

EXINST-1

Method:

EXINST is designed to reconstruct vectors from information written on a binary tape by subroutine XOUT. The user should consult the XOUT write-up for details of method, format, etc.

X, XD, W, and C are arrays. X and XD must be large enough to hold the vector(s) generated. W is used internally by EXINST to store the interpolation coefficients read from tape. It must be dimensioned at least 6K where K equals the total of the dimensions of all vectors on the tape (exclusive of derivatives). C is the critical information array and must be dimensioned at least as large as the corresponding array as used by XOUT.

Usage

There are three distinct calls of EXINST:

- 1) Initialization
- 2) Normal
- 3) Limit shift.

The calls are illustrated in chart form below, followed by a detailed discussion of each.

EXINST-2

IN			OUT	
	Initial	Normal	Limit Shift	
T	Ignored	Time	New Limit	
M	Zero	Vector Number	Non-Zero	M > 0, normal, vector number M = 0, end of data M < 0, vector M not available
N	NC, the dimension of the C array	> 0	< 0	N > 0, normal, vector dimension N = 0, an error condition N < 0, the C array has just been filled (N = -1 tape read backward; N = -2 tape read forward)
X				Vector (if normal return)
XD				Derivative (if available and normal return)
W	Buffer	Buffer	Buffer	
C				Critical information if N < 0
NB	Logical Tape Number	Logical Tape Number	Logical Tape Number	

- 1) Initialization Call: This call must be the first call of EXINST for each case. It initializes certain parameters and reads in the critical information with which each case is started (or terminated if the case is being read in reverse). No vectors are generated by the initialization call.

- 2) Normal Call: This call will normally regenerate the specified vector at the specified time (and its derivative if available). If M and N are returned positive a vector has been generated, otherwise not. If M = 0 an end of the stored data has been reached.

EXINST-3

If M is negative, the subroutine has been unable to find coefficients for the vector M. If N is negative a critical record has been read, i.e., the C array contains new information and no vector has been generated.

3) Limit Shift: Whenever a critical record is read by EXINST (N returned negative), the time associated with this record is stored internally in EXINST. This time is designated TC. Subsequent to a critical event the user may make a limit shift call of EXINST with time argument TN which will then be regarded by EXINST on subsequent normal calls as a new critical time. Essentially this device allows extrapolation at discontinuities occurring at critical points. Thus, if a critical event was preceded by a set of coefficients C1 applying over the interval (T1, TC) and followed by a set of coefficients C2 applying over the interval (TC, T2), a limit shift call of EXINST with time argument TN, where TN is in the interval (T1, TN) and set C2 to apply on the interval (TN, T2). TN was required to be in the interval (T1, T2) merely to make the illustration definite. It is in no way restricted. However, the user should bear in mind that EXINST is being forced to extrapolate, thus values of TN not in the interval (T1, T2) may cause EXINST to give poor results.

For certain applications the user may wish to use directly the information stored on the binary tape. In that instance, the W array may be regarded as output of EXINST. Its format is illustrated below.

	W(1, K)	W(2, K)	W(3, K)	W(4, K)	W(5, K)	W(6, K)
Vector W/Deriv.	M	N	x_1	\dot{x}_1	\ddot{x}_{12}	$\ddot{\ddot{x}}_{12}$
Vector W/O Deriv.	M	-N	x_1	--	x_2	--

EXINST-4

\$IBFTC MC13JY NOREF,M94,NODD,XR3	
CMC13JY GETS VECTORS FROM TAPE SAME AS EXIN EXCEPT FOR COUNTER	XNST0001
SUBROUTINE EXINST(TTT,MMM,NNN,X,XD,W,C,NNB,ITRIG,KOUNT)	XNST0002
C T=TIME M=VECTOR NUMBER N=DIMENSION	XNST0003
C X=VECTOR XD=DERIVATIVE W=W(6,N1+...+NM)=STORAGE	XNST0004
C NB=BINARY TAPE NO. C=C(NCRIT)=CRITICAL INFO BUFFER	XNST0005
C KOUNT IS A LOGICAL RECORD COUNTER WHICH OPERATES WHEN ITRIG = 1	XNST0006
C AND IS USED EXTERNALLY TO CONTROL BACKSPACING FOR TAPES ON WHICH	XNST0007
C THE INDEPENDENT VARIABLE IS INCREASING	XNST0008
DIMENSION X(3),XD(3),W(6,3),C(10),TAY(5)	XNST0009
EQUIVALENCE (EM,MM),(EN,NN)	XNST0010
DOUBLE PRECISION TTT,T,TL,TR,T1,T2,TC	XNST0011
DATA TAY(1),TAY(2)/0.,1./	XNST0012
T=TTT	XNST0013
M=MMM	XNST0014
N=NNN	XNST0015
NB=NNB	XNST0016
100 IF(M.EQ.0) GO TO 180	XNST0017
IF(N.LE.0) GO TO 200	XNST0018
ICRIT=0	XNST0019
110 IF(T.LT.TL) GO TO 160	XNST0020
IF(T.GT.TR) GO TO 170	XNST0021
I=1	XNST0022
115 EM=W(1,I)	XNST0023
EN=W(2,I)	XNST0024
N=IABS(NN)	XNST0025
IF(MM.EQ.M) GO TO 120	XNST0026
I=I+N	XNST0027
IF(I.LE.KMAX) GO TO 115	XNST0028
M=-M	XNST0029
GO TO 900	XNST0030
C COEFFICIENTS FOR VECTOR M HAVE BEEN FOUND	XNST0031
120 IF(NN) 125,900,140	XNST0032
C INTERPOLATE LINEARLY	XNST0033
125 R=(T-T1)/(T2-T1)	XNST0034
DO 130 J=1,N	XNST0035
X(J)=W(3,I)+R*(W(5,I)-W(3,I))	XNST0036
I=I+1	XNST0037
130 CONTINUE	XNST0038
GO TO 900	XNST0039
C USE FULL INTERPOLATION	XNST0040
140 TAY(3)=T-T1	XNST0041
TAY(4)=TAY(3)*TAY(3)/2.	XNST0042
TAY(5)=TAY(4)*TAY(3)/3.	XNST0043
DO 150 K=1,N	XNST0044
X(K)=0.	XNST0045
XD(K)=0.	XNST0046
DO 145 J=1,4	XNST0047
X(K)=X(K)+TAY(J+1)*W(J+2,I)	XNST0048
XD(K)=XD(K)+TAY(J)*W(J+2,I)	XNST0049
145 CONTINUE	XNST0050
I=I+1	XNST0051
150 CONTINUE	XNST0052
GO TO 900	XNST0053
C INITIALIZATION	XNST0054
180 M=1	XNST0055
IDIR=0	XNST0056
NCRIT=N	XNST0057
GO TO 300	XNST0058
C TIME REQUESTED IS BELOW RANGE IN CORE	XNST0059
160 IDIR=-INC	XNST0060
GO TO 300	XNST0061
C TIME REQUESTED IS ABOVE RANGE IN CORE	XNST0062
170 IDIR=INC	XNST0063
GO TO 300	XNST0064
C THIS PATH SHOULD BE TAKEN ONLY WHEN RANGE IS BEING ADJUSTED	XNST0065
200 N=1	XNST0066
ICRIT=1	XNST0067
C DETERMINE RANGE WITH NEW CRITICAL TIME	XNST0068
210 IF(TC-T) 220,900,240	XNST0069
220 IF(IDIR) 300,230,230	XNST0070
230 IF(INC) 270,899,260	XNST0071
240 IF(IDIR) 250,250,300	XNST0072
250 IF(INC) 260,899,270	XNST0073
260 TR=T	XNST0074

GO TO 900	XNST0075
270 TL=T	XNST0076
GO TO 900	XNST0077
C READ IN A SET OF INFORMATION	XNST0078
300 CONTINUE	XNST0079
IF(IDIR) 305,308,308	XNST0080
305 CONTINUE	XNST0081
CALL BACK(NB,4)	XNST0082
IF(ITRIG.EQ.1) KOUNT=KOUNT-4	XNST0083
308 CONTINUE	XNST0084
READ(NB) KEY,IKY,TIME	XNST0085
IF(ITRIG.EQ.1) KOUNT=KOUNT+1	XNST0086
IF(KEY.NE.3) GO TO 400	XNST0087
KMAX=IKY	XNST0088
READ (NB) T1,T2,((W(I,J),I=1,6),J=1,KMAX)	XNST0089
IF(ITRIG.EQ.1) KOUNT=KOUNT+1	XNST0090
995 FORMAT(2018,1P4E18.6)	XNST0091
IF(T1.GT.T2) GO TO 310	XNST0092
TL=T1	XNST0093
TR=T2	XNST0094
INC=1	XNST0095
GO TO 320	XNST0096
C INDEPENDENT VARIABLE IS DECREASING	XNST0097
310 TL=T2	XNST0098
TR=T1	XNST0099
INC=-1	XNST0100
320 CONTINUE	XNST0101
IF(IDIR) 350,900,350	XNST0102
350 IF(ICRIT.LE.0) GO TO 110	XNST0103
IDIR=-IDIR	XNST0104
GO TO 210	XNST0105
C CRITICAL EVENT PROCEDURE	XNST0106
400 IF(KEY-2) 999,405,999	XNST0107
405 N=-1	XNST0108
READ (NB) (C(I),I=1,NCRIT)	XNST0109
IF(ITRIG.EQ.1) KOUNT=KOUNT+1	XNST0110
C IKY=+1 FIRST RECORD OF CASE, IKY=-1 LAST RECORD OF CASE	XNST0111
IF(IKY) 410,430,420	XNST0112
410 CONTINUE	XNST0113
IF(IDIR) 305,305,415	XNST0114
415 M=0	XNST0115
CALL BACK(NB,2)	XNST0116
IF(ITRIG.EQ.1) KOUNT=KOUNT-2	XNST0117
GO TO 890	XNST0118
420 CONTINUE	XNST0119
IF(IDIR) 425,308,308	XNST0120
425 M=0	XNST0121
CALL BACK(NB,3)	XNST0122
IF(ITRIG.EQ.1) KOUNT=KOUNT-3	XNST0123
GO TO 900	XNST0124
C NORMAL CRITICAL EVENT	XNST0125
430 CONTINUE	XNST0126
IF(IDIR) 435,308,440	XNST0127
435 TC=T1	XNST0128
GO TO 900	XNST0129
440 TC=T2	XNST0130
890 CONTINUE	XNST0131
N=-2	XNST0132
900 MMM=M	XNST0133
NNN=N	XNST0134
899 CONTINUE	XNST0135
RETURN	XNST0136
999 CONTINUE	XNST0137
N=0	XNST0138
GO TO 900	XNST0139
END	XNST0140

Subroutine: FIEF

Purpose: To space forward over N files, or backward over
|N| + 1 files.

Calling Sequence: CALL FIEF(N, IER, IB)

Input and Output

I/φ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	N				If N > 0, FIEF spaces forward over N end-of-file marks. If N < 0, FIEF spaces backward and over N + 1 EOF marks then forward over the N +1 st.
φ	IER				IER = 0, success exit. = - 1, end-of-tape encountered. = - 2, read check error.
I	IB				Logical tape number.

Common storages used: None

Subroutines required: None

FIEF-1

\$IBMAP	MC1398		
	ENTRY	FIEF	FIEF0001
	REM	FIEF(N,IER,IB)	FIEF0002
	REM	FIEF SPACES OVER N FILES, N + OR -	FIEF0003
	REM	IGNORES IB, USES ONLY UNIT 10 FOR TRK DAT SIM	FIEF0004
	REM	IER=0, SUCCESS EXIT	FIEF0005
	REM	IER=2, READ ERROR (CHECK SUM)	FIEF0006
FIEF	TXI	FFF,,1	FIEF0007
FFF	SAVE	(1,2)I	FIEF0008
	STZ*	4,4	FIEF0009
	SXA	ER1,4	FIEF0010
	CLA*	FIL	FIEF0011
	STA	FILE	FIEF0012
	CLA*	3,4	FIEF0013
	PAX	0,2	FIEF0014
	TZE	BCKUP	FIEF0015
	TMI	BCKUP	FIEF0016
FILE	TSX	.READ,4	SKIP N FILES
FILE	PZE	**	
	PZE	EOF,,ERR	
	IORPN	0,,0	
	TCH	*-1	
ERR	CLS	=2	IER=-2, READ ERROR
ER1	AXT	*,4	
	STO*	4,4	
	RETURN	FFF	
EOF	TIX	FILE,2,1	
	RETURN	FFF	SUCCESS EXIT, FILE SKIP
BCKUP	TXI	*+1,2,1	BACKSPACE N+1 FILES
	PXD	0,2	
	ORA	FILE	
	SLW	BKP	
	TSX	.BSF,4	
BKP	PZE	0,,0	
	RETURN	FFF	
FIL	PZE	.UN10.	
	END		

Subroutine: FIFL

Purpose: To locate a specified case on the binary tape, and to read record 1 of that case.

Calling Sequence: CALL FIFL (ICASE,J,REM,IER,IB)

Input and Output

I/∅	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	ICASE				Case number. If ICASE = 0, FIFL initializes and does not search.
∅	J				J = REM(2) = number of logical records per set.
∅	REM	(255)			Vector written by TPFL.
∅	IER				IER is set ≠ 0 if case cannot be found.
I	IB				Logical tape number.

Common storages used: None

Subroutines required: FIEF, ERR∅UT

FIFL-1

SIBFTC MC1397 XR3,LIST	
SUBROUTINE FIFL(ICASE,J,IREM,IER,IB)	FIFL000
DIMENSION IREM(255)	FIFL001
C FIFL FINDS AND READS SET 0 OF CASE ICASE.	FIFL002
C ICASE = 0 INITIALIZES AND DOES NOT READ.	FIFL003
C CALLS FIEF TO SPACE OVER FILES TO READING POSITION.	FIFL004
C CALLS ERRROUT IF ERROR FROM FIEF,OR IF 2 TRIES TO LOCATE	FIFL005
C ICASE FAIL.	FIFL006
C ASSUMES CASE NUMBERS SEQUENTIAL AND INCREASING ON TAPE.	FIFL007
C (SEARCHES FORWARD OR BACKWARDS).	FIFL008
IF(ICASE.NE.0) GO TO 50	FIFL009
ICSV =0	FIFL010
RETURN	FIFL011
50 IER=0	FIFL012
ISWT=0	FIFL013
75 CALL FIEF(ICASE-ICSV,IER,IB)	FIFL014
IF(IER.NE.0)GO TO 150	FIFL015
READ(IB)IREM	FIFL016
ICSV= IREM(1)	FIFL017
J=IREM(2)	FIFL018
IF(ICASE.EQ.ICSV) RETURN	FIFL019
ISWT =ISWT+1	FIFL020
GO TO(75,100),ISWT	FIFL021
100 CALL ERRROUT(1,29H(5X,4HCASE,I5,11HNOT ON TAPE),ICASE)	FIFL022
RETURN	FIFL023
150 CALL ERRROUT(1,38H(5X,27HERROR EXIT FROM READ, IER = I2),IER)	FIFL024
RETURN	FIFL025
END	FIFL026

Subroutine: FIST

Purpose: To rewind the binary input tape and to read and check the header (first file). Also initializes subroutine FIFL.

Calling Sequence: CALL FIST (HEAD, IER, IB)

Input and Output

I/φ	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	HEAD	(12)			Header to be checked against the first file of tape. The header from the tape will be in blank common on exit.
φ	IER				IER=2 if HEAD does not agree. Otherwise it is returned unchanged.
I	IB				Logical tape number.

Common storages used: 12 cells

Subroutines required: FIFL

FIST-1

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$IBFTC MC1396 XR3
SUBROUTINE FIST(IHED,IER,IB)
COMMON IR
DIMENSION IHED(12),IR(12)
C FIST REWINDS TAPE IB
C READS HEADER AND CHECKS
C RETURNS IER =2 IF HEADER DOES NOT CHECK
C RETURNS IER UNTOUCHED IF HEADER CHECKS
C INITIALIZES FIFL.
C TAPE HEADER IS IN FIRST 12 CELLS OF BLANK COMMON

REWIND IB
READ(IB) IR
DO 10 I =1,12
IF( IHED(I).NE.IR(I)) GO TO 25
10 CONTINUE
20 CALL FIFL(0,IR,IR,IR,IB)
RETURN
25 IER =2
GO TO 20
END

```

```

FIST000
FIST001
FIST002
FIST003
FIST004
FIST005
FIST006
FIST007
FIST008
FIST009
FIST010
FIST011
FIST012
FIST013
FIST014
FIST015
FIST016
FIST017
FIST018

```

Subroutine: FNØRM

Purpose: Computes the magnitude of a 3-vector.

Calling Sequence: Z = FNØRM (X)

Input and Output

I/Ø	Symbolic Name or Location	Program Dimensions	Math Symbol	Data Dimensions or Units	Definition
I	X	(3)			Input vector
Ø	Z				X

Common storages used: None

Subroutines required: None

FNØRM-1

```
$IBFTC FNORM XR3,M94,NODD
  FUNCTION FNORM(X)
  DIMENSION X(3)
  SUM = 0.
  DO 1 I=1,3
  1 SUM = SUM+X(I)*X(I)
  FNORM = SQRT(SUM)
  RETURN
  END
```

```
FNRM0001
FNRM0002
FNRM0003
FNRM0004
FNRM0005
FNRM0006
FNRM0007
FNRM0008
```

Subroutine: .FVIØ.

Purpose: To obtain IØCS pointer word address for variable tape units. This is a replacement for the IBSYS library routine.

Calling Sequence: None

Common storages used: None

Subroutines required: None

.FVIØ.-1

For read or write statements using variable tape logical numbers, such as

WRITE(N,FORMAT)LIST

the ICS pointer word is obtained by reference to a table in the library routine .FVIO.. The library routine contains pointers (.UNOn.) for all logical units from 1 through (usually) 12, and this results in the attachment of buffers for each unit. By removing entries for all units which are not required from .FVIO., we release the associated buffer space for use by the program.

In the furnished deck, each unused unit in the table is replaced by the dummy name FILE. To add unit n to the table, the symbol .UNOn. replaces the symbol FILE in the nth entry.

.FVIO.-2

\$IBMAP	MCL310	25			
	TTL		FVIO - SET FILE ADDRESS FROM LOGICAL UNIT NO		FVIO
.FVIO.	SAVE				FV100001
	CLA*	3,4	LOGICAL NO		FV100002
	TZE	ERR	NO.EQ.0		FV100003
	PAX	,4			FV100004
	SXD	ARG,4			FV100005
	TXL	OUT,4,MAX			FV100006
ERR	CALL	.FXEM.(ARG)	-LOGICAL UNIT NOT DEFINED FOR VALUE XX-		FV100007
	CALL	EXIT			FV100008
OUT	PAC	,4			FV100009
	CLA*	FILE,4			FV100010
	TZE	ERR	FILE NOT DEFINED		FV100011
	LXA	SYSLOC,4			FV100012
	STO*	4,4	FILE ADDRESS		FV100013
	RETURN	.FVIO.			FV100014
FILE	PZE		FILE NAMES FOLLOW		FV100015
	PZE	FILE	UNIT 1		FV100016
	PZE	FILE			FV100017
	PZE	FILE			FV100018
	PZE	FILE			FV100019
	PZE	.UN05.			FV100020
	PZE	.UN06.			FV100021
	PZE	FILE			FV100022
	PZE	FILE			FV100023
	PZE	FILE			FV100024
	PZE	.UN10.			FV100025
	PZE	.UN11.			FV100026
	PZE	.UN12.			FV100027
ARG	PZE	47			FV100028
MAX	SYN	ARG-FILE-1			FV100029
	END				

REFERENCES

1. "Program Description and Theoretical Basis for the Orbit Determination Program," Philco-Ford Corporation, TR-DA1508, Palo Alto, California, December 1967.
2. "Input-Output Summary for the Orbit Determination Program," Philco-Ford Corporation, TR-DA1510, Palo Alto, California, December 1967.