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COMPUTER PROGRAM FOR PLANETARY SOLAR ARRAY DEVELOPMENT

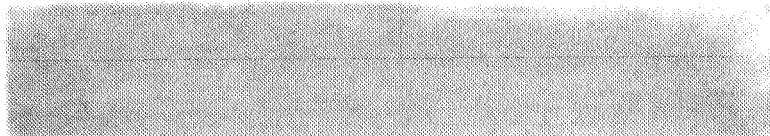
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ABSTRACT

This summary report, submitted under the terms of contract 952606 (subcontract under NASA Contract NAS7-100, Task Order No. RD-26), describes a computer program that will print the power output of a solar panel located arbitrarily on the surface of Mars. The original program was developed under Contract 952035 between the Jet Propulsion Laboratory and Electro-Optical Systems, and is detailed in EOS Report 7254-Final, dated 28 February 1969.

The modifications made under the present contract were to:

- a. Provide for solar cell model input in terms of 200 (or fewer) points describing the room-temperature I-V characteristics, and up to 25 points for each temperature coefficient.
- b. Add a subroutine to define the attenuation and scattering as a result of atmospheric dust, including layering on the panel.
- c. Print out important cell I-V values during three periods of the Martian day.
- d. Perform a brief study of the sand dune height expected on Mars.

Theoretical calculations of Rayleigh and Mie scattering are derived in the report.

The program is included; it is written in Fortran IV for an XDS Sigma VII computer. Instructions are given to prepare the set of necessary input parameters; an explanation and sample printout are included.

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SECTION 1

INTRODUCTION

Included in this report are the following major items:

- a. A theoretical derivation treating the Mie and Rayleigh scattering due to the molecules and particles present in the Martian atmosphere.
- b. A theoretical derivation of the expected dust layer buildup on a solar panel, by means of the Stokes-Cunningham equation, and a calculation of light attenuation through this layer.
- c. A theoretical study of the parameters of the sand dunes expected on the Martian surface, based upon the Bagnold equations, and other physical characteristics of the surface environment.
- d. A discussion of a solar cell model and description of its computer implementation.
- e. The computer program, typical printouts, and a set of recommendations.

SECTION 2

THEORETICAL SCATTERING AND DUST LAYERING

2.1 GENERAL

In order to determine the output of a solar cell on the Martian surface, it is necessary to know the intensity of the sunlight at the location of the solar cell on the planet's surface. Sunlight which enters the Martian atmosphere is both absorbed and scattered by atmospheric molecules. Dust particles suspended in the atmosphere will also absorb and scatter the sunlight, but, for the sake of simplicity, we shall consider the dust particles to be perfect dielectrics (i.e., nonconducting), so that they do not absorb the solar energy. The intensity of the sunlight at the solar panel depends on the absorption coefficient of the molecular atmosphere and the scattering coefficient of both the atmospheric molecules and the dust particles. These coefficients are discussed in the following subsections.

2.2 ABSORPTION COEFFICIENT OF MARTIAN ATMOSPHERE

Let z denote the altitude (vertical distance above the Martian surface) of a point in the Martian atmosphere, and let $\mu(z)$ denote the absorption coefficient of the atmosphere at this point. Then, if I_0 is the intensity of sunlight in free space at Mars' distance from the sun, the intensity of light at a point on the Martian surface directly beneath the sun, resulting only from the loss of energy due to absorption by the atmosphere, is given by

$$I_a = I_0 \exp \left[- \int_0^H \mu(z) dz \right],$$

where H = height of effective atmosphere. Now, the absorption coefficient, $\mu(z)$, is directly proportional to the density, $\rho(z)$, of the atmosphere. Thus,

$$\mu(z) = k \rho(z),$$

where

$$\rho(z) = \rho_o \exp \left[\frac{-\alpha z}{R(R+z)} \right].$$

Here, ρ_o = density of atmosphere at the surface of Mars = 1.465×10^{-5} gm/cm³, R = radius of Mars = 3332 km, and

$$\alpha = \frac{K\rho_o}{p_o}$$

where p_o = atmospheric pressure at the surface of Mars = 7.0×10^3 dynes/cm² and K = gravitational constant of Mars = 4.255×10^{19} cm³/sec². Here, $\alpha = 8.9 \times 10^5$ km. Combining the first three equations gives

$$\frac{I_a}{I_o} = \exp \left\{ -k\rho_o \int_0^H \exp \left[\frac{-\alpha z}{R(R+z)} \right] dz \right\},$$

for which

$$k = \frac{\ln(I_o/I_a)}{\rho_o \int_0^H \exp \left[\frac{-\alpha z}{R(R+z)} \right] dz}.$$

Therefore we finally have

$$\mu(z) = k\rho(z) = \frac{\ln(I_o/I_a)}{\int_0^H \exp \left[\frac{-\alpha z}{R(R+z)} \right] dz} \cdot \exp \left[\frac{-\alpha z}{R(R+z)} \right]. \quad (1)$$

The values of ρ_0 , p_0 given above are taken from Ref. 1. For the purpose of our problem, we shall let $I_a/I_0 = 0.95$, so that $\ln(I_0/I_a) = 0.05126$. Also, the height of the effective Martian atmosphere is taken as $H = 200$ km. A computer calculation yields

$$\int_0^H \exp \left[\frac{-\alpha z}{R(R+z)} \right] dz = 12.57 \text{ km.}$$

2.3 RAYLEIGH SCATTERING COEFFICIENT

According to Ref. 1 (Fig. 4-25), the maximum solar irradiance on the Martian surface occurs at a wavelength $\lambda = 0.5\mu$ ($\mu = 1$ micron $= 10^{-4}$ cm). Since the diameter of an atmospheric molecule is considerably less than 0.1λ , the scattering of the sunlight by the atmospheric molecules will be Rayleigh-type scattering. The Rayleigh scattering coefficient, s_R , is given by

$$s_R = n_R \sigma_R,$$

where n_R = molecular number density of the Martian atmosphere and σ_R = Rayleigh scattering cross section. Here, $n_R = \rho/m$, where m = mass of an atmospheric molecule. From Ref. 1, we deduce that $m = 6.3 \times 10^{-23}$ gram. Therefore,

$$n_R = n_0 \exp \left[\frac{-\alpha z}{R(R+z)} \right],$$

where n_0 = molecular number density at the surface of Mars $= 2.32 \times 10^{17} \text{ cm}^{-3}$.

According to Ref. 2, p.3, the Rayleigh scattering cross section is given by

$$\sigma_R(\lambda) = \frac{8 \pi^3 (m_s^2 - 1)^2}{3 \lambda^4 n_o^2} \cdot \frac{6 + 3\delta}{6 - 7\delta},$$

where m_s = index of refraction of Mars' atmosphere at the surface of Mars and δ = depolarization factor. It can be shown that $m_s = 1.000004$ and $\delta = 0.075$. Then, since $\lambda = 0.5\mu$, we have $\sigma_R = 1.79 \times 10^{-26} \text{ cm}^2$.

Consequently,

$$s_R = n_o \sigma_R \exp \left[\frac{-\alpha z}{R(R+z)} \right], \quad (2)$$

where $n_o \sigma_R = 4.15 \times 10^{-4} \text{ km}^{-1}$.

2.4 MIE SCATTERING COEFFICIENT

The diameters of the dust particles in the Martian atmosphere being considered for this problem lie in the range 1.0μ , 100.0μ . Since these diameters are the same order of magnitude as $\lambda = 0.5\mu$, and greater, the scattering of the sunlight by the dust particles will be Mie-type scattering. The classical theory which describes this type of scattering assumes that the scattering particles are spherical. Also as mentioned in the introduction, we shall assume, for simplicity, that the dust particles are nonconducting. The Mie scattering coefficient, s , is given by

$$s = n\sigma, \quad (3)$$

where n = number density (constant) of dust particles in the atmosphere, and σ is the Mie scattering cross section. According to Ref. 3,

$$\sigma(\lambda, N, \rho) = \frac{\lambda^2}{2\pi} \sum_{i=1}^{\infty} (2i+1) (|a_i|^2 + |b_i|^2), \quad (4)$$

where N = index of refraction of a dust particle, and

$$\rho = \frac{2\pi a}{\lambda},$$

where a = radius of dust particle. Here,

$$a_i = \frac{N\psi_i(\rho)\psi_i'(\rho) - \psi_i'(\rho)\psi_i(\rho)}{\psi_i(\rho)\zeta_i'(\rho) - N\psi_i'(\rho)\zeta_i(\rho)}$$

and

$$b_i = \frac{N\psi_i'(\rho)\psi_i(\rho) - \psi_i(\rho)\psi_i'(\rho)}{\psi_i'(\rho)\zeta_i(\rho) - N\psi_i(\rho)\zeta_i'(\rho)},$$

where the primes denote differentiation with respect to the argument of the respective function. Further,

$$\psi_i(x) = \left(\frac{\pi x}{2}\right)^{1/2} J_{i+1/2}(x)$$

and

$$\zeta_i(x) = \psi_i(x) - j(-1)^i \left(\frac{\pi x}{2}\right)^{1/2} J_{-(i+1/2)}(x),$$

where $j = (-1)^{1/2}$.

Also,

$$J_{i+1/2}(x) = \left(\frac{2}{\pi x}\right)^{1/2} \left[P_i(x) \sin\left(x - \frac{i\pi}{2}\right) + Q_i(x) \cos\left(x - \frac{i\pi}{2}\right) \right];$$

$$J_{-(i+1/2)}(x) = \left(\frac{2}{\pi x}\right)^{1/2} \left[P_i(x) \cos\left(x + \frac{i\pi}{2}\right) - Q_i(x) \sin\left(x + \frac{i\pi}{2}\right) \right].$$

Here,

$$P_i(x) = \sum_{p=0}^{\leq \frac{i}{2}} \frac{(-1)^p (i+2p)!}{(2p)!(i-2p)!} \frac{1}{(2x)^{2p}}$$

and

$$Q_i(x) = \sum_{p=0}^{\leq \frac{1}{2}(i-1)} \frac{(-1)^p (i+2p+1)!}{(2p+1)!(i-2p-1)!} \frac{1}{(2x)^{2p+1}},$$

where the upper limit that ultimately belongs on each summation sign is the nearest integer to the value indicated. The quantities $J_{i+1/2}$ and $J_{-(i+1/2)}$ are the Bessel Functions of half-integral order.

2.5 INTENSITY OF SUNLIGHT AT THE SOLAR PANEL

The total intensity of sunlight at the solar panel on the Martian surface consists of three parts: (1) the intensity, I_1 , resulting from the attenuation of solar energy due to absorption (by the atmospheric molecules) and scattering (by the molecules and dust particles); (2) the intensity, I_{sm} , due to energy scattered by the molecules; (3) the intensity, I_{sd} , due to energy scattered by the dust particles. Thus, the total intensity of sunlight at the solar panel is

$$I_T = I_1 + I_{sm} + I_{sd} \quad (5)$$

2.5.1 INTENSITY, I_1 , DUE TO ATTENUATION

Consider the rays of light from the sun to be incident on the solar panel at an arbitrary angle, γ , relative to the local vertical, as shown in Fig. 1. When $\gamma = 0^\circ$, the sun is directly over the panel, and when $\gamma = 90^\circ$, the sun is on the horizon. The intensity I_1 , for arbitrary γ , is given by

$$I_1 = I_o \exp \left\{ - \int_0^L [\mu(\xi) + s_R(\xi) + s(\xi)] d\xi \right\}, \quad (6)$$

where

$$\begin{aligned} & \int_0^L [\mu(\xi) + s_R(\xi) + s(\xi)] d\xi \\ & \approx \left[\frac{\ln(I_o/I_a)}{\int_0^H \exp\left[\frac{-\alpha z}{R(R+z)}\right] dz} + n_o \sigma_R \right] \\ & \cdot \int_0^L \exp\left[-\frac{\alpha}{R} \left(\frac{1}{2} \frac{\xi^2}{R^2} + \frac{\xi \cos \gamma}{R} \right)\right] d\xi + sL, \end{aligned} \quad (7)$$

and

$$L = -R \cos \gamma + (H + R) \left[1 - \left(\frac{R \sin \gamma}{H + R} \right)^2 \right]^{1/2}.$$

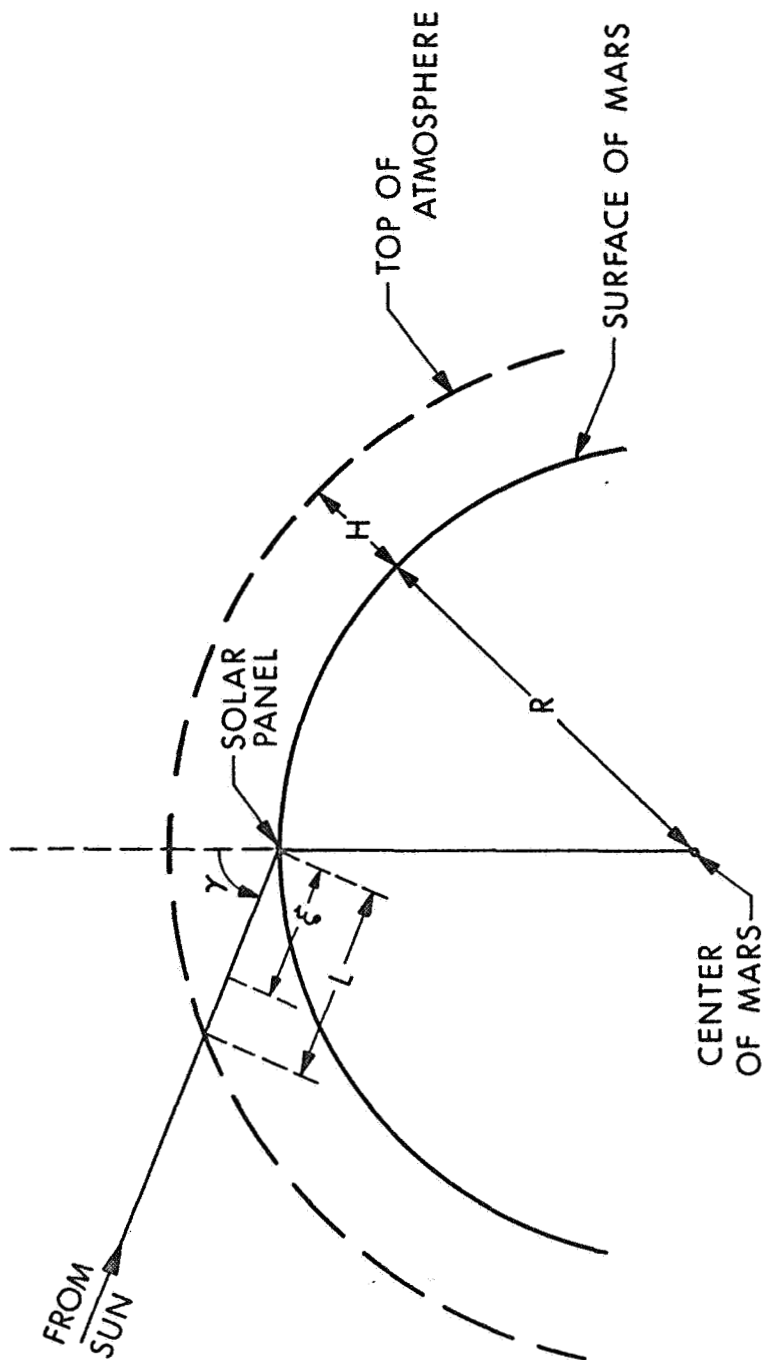


Figure 1. Definition of γ

A digital computer is needed to evaluate the above integral. However, in the special cases when $\gamma = 0^\circ$ or 90° , the integration may be performed analytically. Thus, when $\gamma = 0^\circ$,

$$\begin{aligned} \int_0^L \exp\left[-\frac{\alpha}{R} \left(\frac{1}{2} \frac{\xi^2}{R^2} + \frac{\xi \cos \gamma}{R}\right)\right] d\xi &= \int_0^H \exp\left[-\frac{\alpha}{R} \left(\frac{1}{2} \frac{\xi^2}{R^2} + \frac{\xi}{R}\right)\right] d\xi \\ &\approx \int_0^H \exp\left(-\frac{\alpha}{R} \cdot \frac{\xi}{R}\right) d\xi \approx \frac{R^2}{\alpha} = 12.48 \text{ km}, \end{aligned} \quad (8)$$

and when $\gamma = 90^\circ$,

$$\begin{aligned} \int_0^L \exp\left[-\frac{\alpha}{R} \left(\frac{1}{2} \frac{\xi^2}{R^2} + \frac{\xi \cos \gamma}{R}\right)\right] d\xi &= \int_0^L \exp\left(-\frac{\alpha}{R} \cdot \frac{1}{2} \frac{\xi^2}{R^2}\right) d\xi \\ &\approx \int_0^\infty \exp\left(-\frac{\alpha}{R} \cdot \frac{1}{2} \frac{\xi^2}{R^2}\right) d\xi = \left(\frac{\pi R^3}{2\alpha}\right)^{1/2} = 256 \text{ km}, \end{aligned}$$

where $L = (H^2 + 2HR)^{1/2} = 1170 \text{ km}$.

2.5.2 INTENSITY, I_{sm} , DUE TO SCATTERING BY MOLECULES

In the derivation of I_{sm} , and also I_{sd} , it is assumed that the scattered sunlight reaching the solar panel has been scattered only once, i.e., multiple scattering is not taken into account. For this reason, and because of the limited region of space in which the scattered light is assumed to originate (explained below and in Fig. 2), the intensity of the scattered sunlight at the solar panel will represent a conservative value.

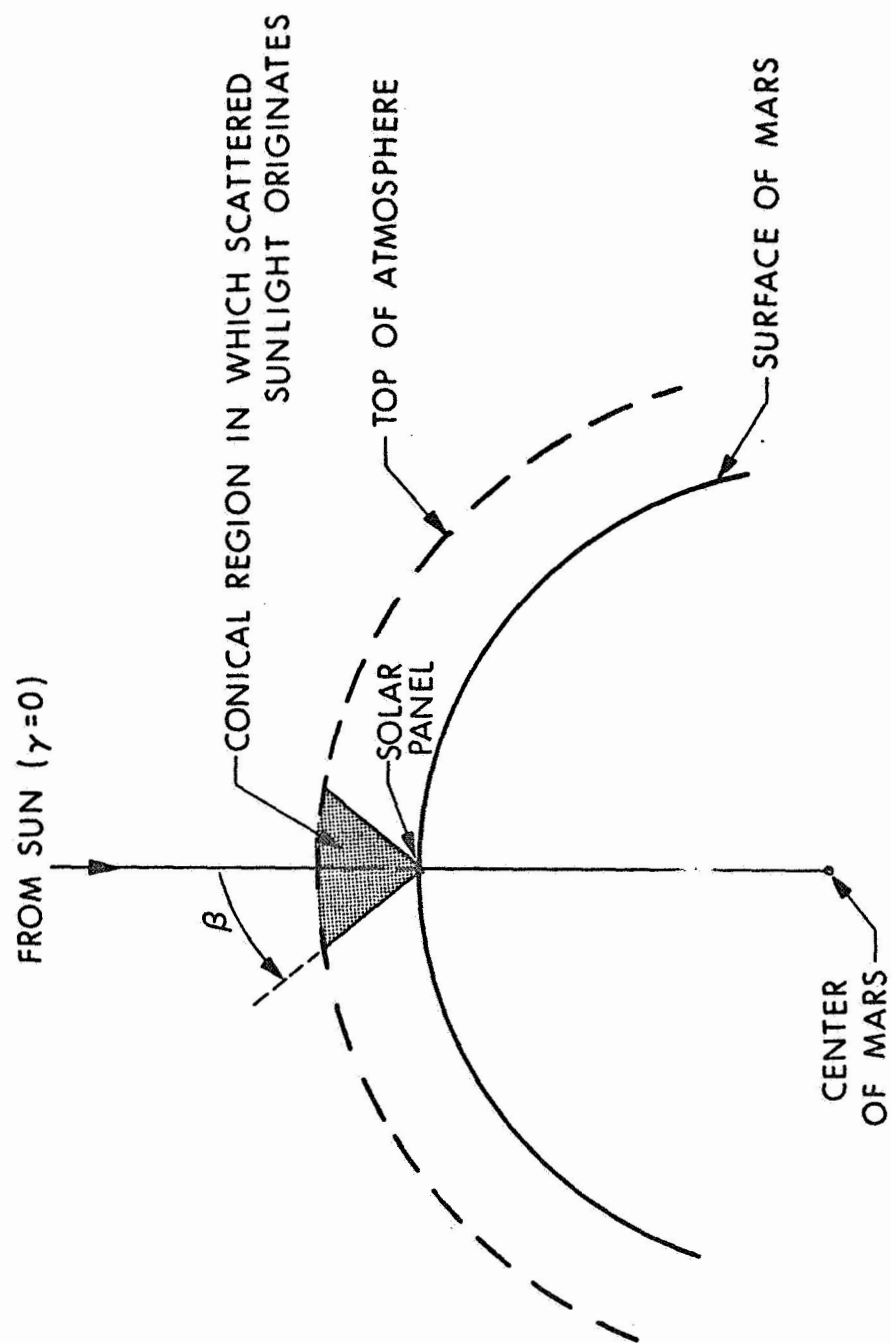


Figure 2. Definition of β

The derivation of I_{sm} for arbitrary γ is rather difficult. For the purpose of this study, only the I_{sm} corresponding to $\gamma = 0^\circ$ will be derived. Preliminary calculations of I_{sm} , in which it is assumed that the scattered light reaching the solar panel originates in a long cylindrical column whose axis coincides with the panel-sun direction and whose cross section is the projected area of the solar panel surface, show that the ratio, I_{sm}/I_1 , is the same for $\gamma = 0^\circ$ and $\gamma = 90^\circ$. We shall assume that the ratio I_{sm}/I_1 remains constant for all values of γ ($0^\circ \leq \gamma \leq 90^\circ$). Hence, the value of I_{sm} for an arbitrary γ can be obtained from,

$$I_{sm} = \frac{(I_{sm})_{\gamma=0}}{(I_1)_{\gamma=0}} \cdot I_1, \quad (10)$$

where $(I_1)_{\gamma=0}$ is obtained from Eqs. 6, 7, 8 and I_1 is determined from Eqs. 6 and 7. In order to reduce the mathematical complexity in the derivation of $(I_{sm})_{\gamma=0}$, it is assumed that the scattered (by the atmospheric molecules) sunlight incident on the solar panel comes only from that part of the atmosphere contained in a cone of half-angle β , with vertex at the solar panel and central axis coincident with the local vertical. The shaded area in Fig. 2 denotes this conical region. Here, $\beta \leq 45^\circ$. The intensity is given by

$$\begin{aligned} (I_{sm})_{\gamma=0} = & \frac{3}{8} I_o n_o \sigma_R e^{-sH} \int_0^\beta \frac{(1 - \eta \sec\theta)(1 + \cos^2\theta)\sin\theta}{\frac{\alpha}{R^2} \cos\theta + s(1 - \cos\theta)} d\theta \\ & + \frac{3}{8} I_o \eta n_o \sigma_R e^{-sH} \int_0^\beta \frac{(1 + \cos^2\theta)\tan\theta}{2\frac{\alpha}{R^2} \cos\theta + s(1 - \cos\theta)} d\theta, \end{aligned} \quad (11)$$

where

$$\eta = \frac{R^2}{\alpha} \left[\frac{\ln(I_o/I_a)}{\int_0^H \exp\left[\frac{-\alpha z}{R(R+z)}\right] dz} + n_o \sigma_R \right] = 0.0560.$$

The smaller the value of β , the more accurate is Eq. 11. The value of β should not exceed 45° .

2.5.3 INTENSITY, I_{sd} , DUE TO SCATTERING BY DUST PARTICLES

The derivation of I_{sd} for arbitrary γ is also rather difficult, and so, for the purpose of this study, only the I_{sd} corresponding to $\gamma = 0^\circ$ will be derived. Preliminary calculations of I_{sd} , in which it is assumed that the scattered light reaching the solar panel originates in the same cylindrical column as previously described for I_{sm} , show that the ratio I_{sd}/I_1 is the same for $\gamma = 0^\circ$ and $\gamma = 90^\circ$. We shall assume that this ratio is constant for all values of γ ($0^\circ \leq \gamma \leq 90^\circ$), so that the value of I_{sd} for an arbitrary γ can therefore be obtained from

$$I_{sd} = \frac{(I_{sd})_{\gamma=0}}{(I_1)_{\gamma=0}} \cdot I_1. \quad (12)$$

In order to reduce the mathematical complexity in the derivation of $(I_{sd})_{\gamma=0}$, it is assumed that the scattered (by the dust particles suspended in the atmosphere) sunlight incident on the solar panel comes only from the conical region shown in Fig. 2. The intensity is given by

$$\begin{aligned}
(I_{sd})_{\gamma=0} &= \frac{\lambda^2 I_o}{4\pi \sigma} e^{-sH} \int_0^\beta (1 - \eta \sec \theta) \\
&\cdot \left[\frac{1 - \exp \left[-s(1 - \cos \theta) \left(\frac{2RH + H^2}{2R \cos \theta} \right) \right]}{1 - \cos \theta} \right] \\
&\cdot [i_1(\theta) + i_2(\theta)] \sin \theta \, d\theta \\
&+ \frac{\lambda^2 I_o}{4\pi} \eta_n e^{-sH} \int_0^\beta \frac{\tan \theta}{\frac{\alpha}{2} \cos \theta + s(1 - \cos \theta)} \\
&\cdot [i_1(\theta) + i_2(\theta)] \, d\theta, \tag{13}
\end{aligned}$$

where

$$i_1(\theta) = |s_1|^2, \quad i_2(\theta) = |s_2|^2, \tag{14}$$

$$s_1 = \sum_{i=1}^{\infty} \frac{2i+1}{i(i+1)} \left[a_i \frac{P_i^1(\cos \theta)}{d\theta} + b_i \frac{P_i^1(\cos \theta)}{\sin \theta} \right],$$

and

$$s_2 = \sum_{i=1}^{\infty} \frac{2i+1}{i(i+1)} \left[a_i \frac{P_i^1(\cos \theta)}{\sin \theta} + b_i \frac{dP_i^1(\cos \theta)}{d\theta} \right].$$

Here, $P_i^1(\cos \theta)$ are the Associated Legendre Functions, and a_i, b_i are given in subsection 2.4. The dimensionless quantities $i_1(\theta)$, $i_2(\theta)$ are called the intensity functions, or angular distribution functions, of the scattered radiation. Again, the angle β should not exceed 45° . Of course, the same value of β should be used in equations 11 and 13.

It should be noted that a singularity does not exist in the first integral of equation 13. Thus, when $\theta = 0$,

$$\frac{1 - \exp \left[-s(1 - \cos \theta) \left(\frac{2RH + H^2}{2R \cos \theta} \right) \right]}{1 - \cos \theta} = s \left(1 + \frac{H}{2R} \right) H .$$

2.6 EFFECTIVE LIGHT INTENSITY AT THE SOLAR PANEL

The power generated by a solar panel due to a particular ray of light incident on the panel is proportional to the intensity of this ray (at the panel) multiplied by the cosine of the angle between the normal to the panel and the direction of the ray. Since the scattered light reaching the panel arrives from several directions in space (angular variation = 2β), not all of this intensity, I_T , given by equation 5 will be equally effective in generating power. Since the rays of light contributing to the nonscattered intensity, I_1 , come directly from the sun, this intensity should be multiplied by $\cos \chi$, where χ = angle between the normal to the panel and the panel-sun direction. Also, each of the several rays of light contributing to the scattered intensity, $I_{sm} + I_{sd}$, should be multiplied by the appropriate quantity. This latter task is very complicated mathematically. Since the rays of light having the highest intensity are generally those coming directly from the sun, little accuracy will be lost if we multiply the total intensity, I_T , by $\cos \chi$ in order to obtain the power output of the solar panel.

The quantity, $\cos \chi$, is obtained from the computer program described on pages 56 - 68 of EOS Report 7254-Final, "Planetary Solar Array Development," dated 28 February 1969.

Multiplying $I_{sm} + I_{sd}$ by $\cos \chi$ (as recommended above) will give a slightly higher intensity than the product of the scattered intensity constituents and their associated cosines. Some of this additional intensity will be supplied, realistically, by the intensity resulting from multiple scattering and from the intensity due to scattered light originating outside of the conical region defined in Fig. 2.

2.7 ATTENUATION OF SUNLIGHT BY DUST LAYER ON SOLAR PANEL SURFACE

In addition to being attenuated by absorption and scattering in the Martian atmosphere, the incident sunlight will also be attenuated by the layer of dust which is expected to collect on the solar panel surface. Evidence obtained from close-up photographs of Mars suggests the existence of dust storms in the Martian atmosphere. After a dust storm has passed over an area, it is reasonable to expect that the dust particles suspended in the atmosphere will fall to the surface, thereby forming a layer of dust on the solar panel. Of course, a minimal dust layer will probably exist on the solar panel surface while the wind is blowing, but the thickness of this layer will depend largely on the electrostatic forces between the dust particles and the panel surface, and this phenomenon is too complex to be analyzed in this report.

For simplicity, we assume that the dust particles are spherically shaped. According to the Stokes-Cunningham equation (Ref. 4, pp. 592, 593), a small spherical particle falling under gravitation through an atmosphere will reach a constant terminal velocity, v , given by

$$v = \frac{2ga^2}{9\eta} (\rho - \rho_o) \left(1 + \frac{B}{ap_o}\right), \quad (15)$$

where ρ = density of particle, g = gravitational acceleration, a = radius of particle, η = coefficient of viscosity of Martian atmosphere, $B = 0.004632$, and ρ_0 , p_0 are the density and pressure, respectively, of the Martian atmosphere at the surface of Mars. In equation 15, a is expressed in centimeters and p_0 is given in millimeters of mercury. Here, $p_0 = 13.5$ mm Hg.

The thickest dust layer will form on the solar panel if it is lying flat on the Martian surface, i.e., if the normal to the panel lies along the local vertical. If n denotes the number density of dust particles in the Martian atmosphere, then the number of dust particles which will fall on unit area of the solar panel in unit time, i.e., the deposition or settling rate, v , is

$$v = nv, \quad (16)$$

where v is given by equation 15. If A denotes the area of the solar panel surface, then the number of dust particles which are required to completely cover the surface and thereby form a single layer on the solar panel is approximately $A/\pi a^2$. If Δt denotes the time taken for a single layer to form on the surface, then $A/\pi a^2 = v A \Delta t$, or

$$\Delta t = \frac{1}{\pi a^2 v}.$$

We assume that the dust layer thickness increases continuously and uniformly with time. Let h denote the thickness of the layer and let t denote time. Then, since the thickness of a single layer equals $2a$, the general thickness will be $h = (2a/\Delta t)t$, or

$$h = 2\pi a^3 vt, \quad (17)$$

where $t > \Delta t$ and v is given by equation 16.

Values must be assigned to ρ , g , and η before h can be computed. According to Ref. 5, the density of the top few centimeters of the Martian surface is estimated to lie between 0.6 and 2.9 gm/cm³. We shall assume that the density, ρ , of a dust particle in the Martian atmosphere is $\rho = 2.0$ gm/cm³. The gravitational acceleration at the surface of Mars is $g = 382$ cm/sec². An approximate value of η can be determined from information contained in chapter 4 of Ref. 6. We assume that the Martian atmosphere is 100% carbon dioxide and that the temperature of the atmosphere near the Martian surface is 220°K (Ref. 1). The resulting coefficient of viscosity is, from Ref. 6,

$$\eta = 1.11 \times 10^{-4} \text{ gm cm}^{-1} \text{ sec}^{-1} .$$

Combining equations 15, 16, and 17 and employing the above values of ρ , g , and η , we obtain

$$h = 0.96 \times 10^7 \left(1 + \frac{0.343 \times 10^{-3}}{a}\right) a^5 n t , \quad (18)$$

where $t \leq \tau$, $[a] = \text{cm}$, $[n] = \text{cm}^{-3}$, $[t] = \text{sec}$, $[h] = \text{cm}$, and τ denotes the length of time between dust storms. For given values of a and n , the maximum h occurs when $t = \tau$. (It is assumed that, while the wind is blowing, no dust exists on the solar panel, but as soon as the wind stops, dust begins to fall on the panel surface and continues to do so until the next storm at which time all the dust is blown off the panel.) Since the minimum values of a and n considered in our problem are 5×10^{-5} cm and 1 cm^{-3} , respectively, the minimum h , as a function of time, is

$$h_{\min} = 2.36 \times 10^{-14} t ,$$

and since the maximum values of a and n considered in our problem are 5×10^{-3} cm and 10^2 cm^{-3} , respectively, the maximum h , as a function

of time, is

$$h_{\max} = 3.2 \times 10^{-3} t.$$

Note that for the minimum values of a and n , it would take approximately 10^5 years to accumulate a 1 mm thick dust layer, whereas, for the maximum values of a and n , this same thickness would be generated in only 31.2 seconds.

Sunlight incident on the solar panel will be reflected, absorbed, and transmitted by the dust layer on the panel surface. If we let ζ denote the absorption coefficient of the dust layer and let ν denote its reflectivity, then the transmitted intensity, I_2 , which ultimately determines the output power of the solar panel is given by

$$I_2 = I_T \cos \chi (1 - \nu) e^{-\zeta (h \sec \chi)}, \quad (19)$$

where I_T is given by equation 5, h is given by equation 18, and χ is discussed in subsection 2.6. The reflectivity, ν , of the dust layer will not be constant but will depend on the layer thickness, h . A realistic functional dependency is

$$\nu = \nu_0 (1 - e^{-kh}),$$

where ν_0 = reflectivity of dust layer for large values of h . We assume that when $h = 4a$ (a = radius of dust particle), then $\nu/\nu_0 = 0.99$. Hence, $k = 1.15/a \approx 1/a$, so that

$$\nu = \nu_0 (1 - e^{-h/a}), \quad (20)$$

where h is given by equation 18.

Values must be assigned to the parameters ζ and ν_0 . An approximate range of values for ζ is $1 \text{ cm}^{-1} \leq \zeta \leq 100 \text{ cm}^{-1}$. Also, $0 \leq \nu_0 \leq 1$. As reliable estimates of ζ and ν_0 (and k) are obtained from the Mariner missions, they can be substituted in equations 19 and 20 and a correspondingly realistic value of I_2 can then be determined.

SECTION 3

DUNE MOVEMENT STUDY

The existence of sand dunes on the Martian surface can be dangerous to an unmanned Martian mission in two ways: (1) the spacecraft may land on the side of a dune and then sink in the sand and/or topple over; (2) the spacecraft may land near a sand dune and eventually be covered by the dune as the dune is slowly "pushed" along the ground by a steady Martian wind. Neither of these mishaps can be prevented in the case of an unmanned landing, unless the spacecraft lands on a part of Mars which is free of strong winds and dust storms, and hence, free of sand dunes. According to Ref. 5, most of the dust storms on Mars occur in the $0^{\circ} \pm 35^{\circ}$ latitude belt.

If the height of the sand dunes is small compared with the vertical dimension of the landing spacecraft, then either of the above two mishaps will be less likely to occur. It is desirable, therefore, to know the height of sand dunes on the Martian surface. According to Ref. 7, it is not possible to divorce the height of a sand dune from the rate at which the sand is transported across the dune, since the dune would not exist without the surface winds which create it by moving sand particles over the Martian surface. Hence, in order to predict the dune height, one must also make some assumptions regarding the rate of sand movement.

The motion of sand from one side of the dune to the other side has the net effect of displacing the sand dune (very slowly) along the ground in the direction of the wind. According to Ref. 7, the speed, c , at which the sand dune moves along the ground, and the rate, q , at which sand is carried over the dune along a path of unit width, are related to the sand dune height, H , by

$$H = \frac{q}{\gamma c}, \quad (1)$$

where γ = bulk density of the sand dune. Here, $[q]$ = mass/(time x distance). Also according to Ref. 7, the quantity q is given by

$$q = C \left[\frac{0.174}{\log_{10} \frac{z}{k'}} \right]^3 \left(\frac{d}{D} \right)^{1/2} \frac{\rho_o}{g} (v - V_t)^3. \quad (2)$$

Here, C = dimensionless constant which depends on the type of sand, z = height above ground that the wind velocity, v , is measured, k' = height of ripples of sand on ground, d = diameter of sand particles, D = diameter of typical (earth) sand particle = 0.25 mm, ρ_o = density of Martian atmosphere near the surface, g - gravitational acceleration at Mars' surface, v = velocity of wind at a height z , V_t = threshold velocity, i.e., the velocity of wind at height k' at which sand particles on the ground first begin to move. By combining Eqs. 1 and 2 and assigning values to γ , C , z , k' , d , ρ_o , g , and V_t , we will obtain H as a function of c and v . Treating c as a parameter, we can therefore obtain graphs of H versus v , for selected values of c .

According to Ref. 5, the density of the top few centimeters of the Martian surface brackets the value 2.0 gm/cm^3 , which we take for γ . From Ref. 7, $C = 1.8$. Reasonable values to assume for the Martian surface are: $z = 1$ meter, $k' = 10^{-2}$ meters, $d = D$. Also, $\rho_o = 14.65 \times 10^{-6} \text{ gm/cm}^3$, $g = 382 \text{ cm/sec}^2$. Approximate values of V_t as a function of the particle radius, $d/2$, are given in the graph on page 6 of Ref. 8, for two surface pressures of the Martian atmosphere. We consider a 7-millibar air pressure ($p_o = 7.0 \times 10^3 \text{ dynes/cm}^2$) near the surface of Mars. Then, for $d/2 = 0.25 \text{ mm}/2 = 125 \text{ microns}$ and $p = 7 \text{ mb}$, the graph gives $V_t \approx 3.6 \times 10^2 \text{ cm/sec}$. Combining Eqs. 1 and 2 and substituting the above values finally gives

$$H = \frac{2.275 \times 10^{-11}}{c} (v-360)^3, \quad (3)$$

where $[c] = \text{cm/sec}$, $[v] = \text{cm/sec}$, $[H] = \text{cm}$, and $v \geq 360 \text{ cm/sec}$. Figure 3 shows a graph of H versus v for the five values, $c = 0.1, 0.5, 1.0, 5.0$, and 10 cm/sec .

It is interesting to note that, according to Ref. 7, sand dunes on the earth may reach a height of 30 meters and move along the ground at speeds ranging from 0.1 to 0.2 cm/hr.

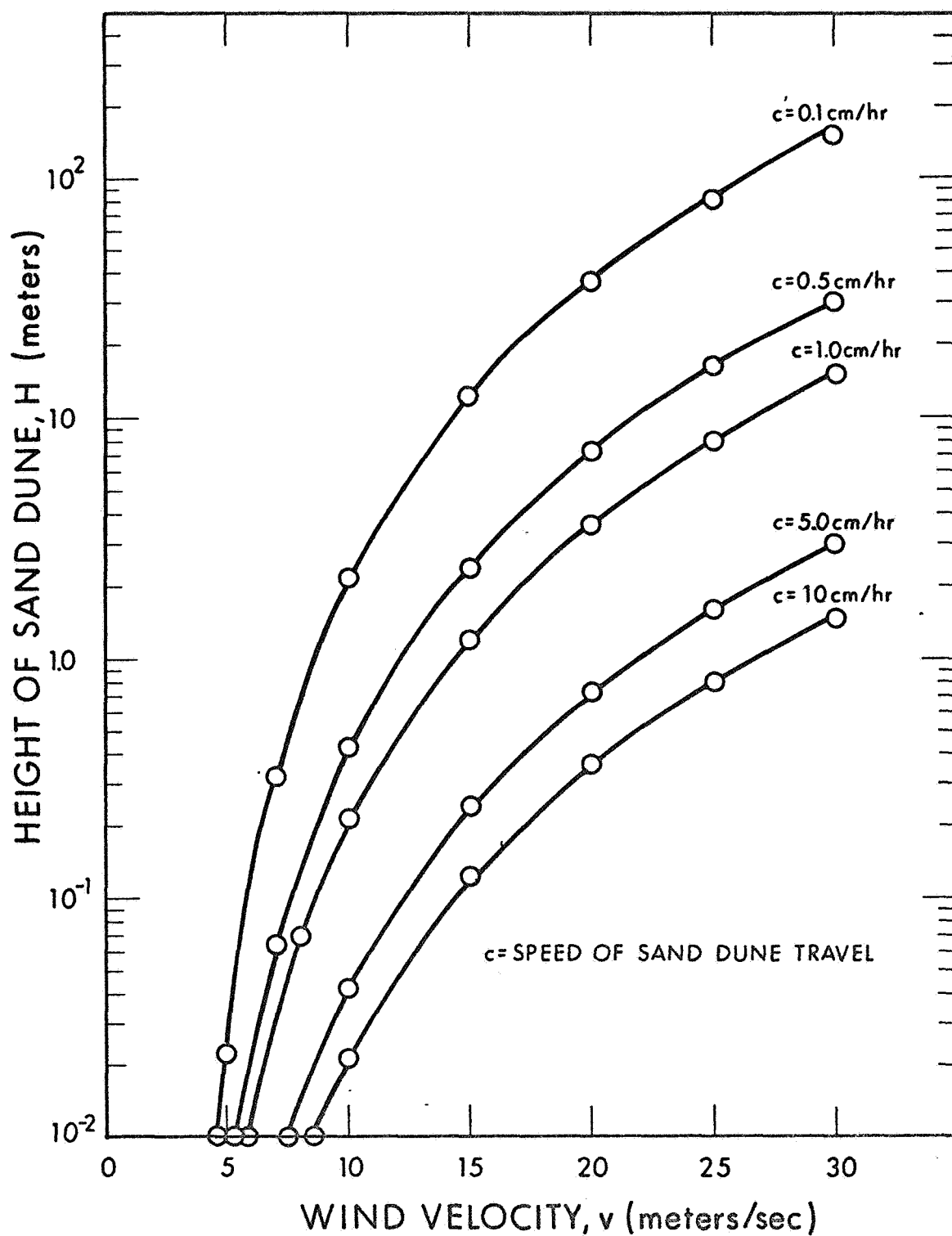


Figure 3. Sand Dune Height versus Wind Velocity

SECTION 4

COMPUTER CELL MODEL PROGRAM ADDITIONS AND CHANGES

Tasks I and II of the present contract were essentially concerned with modifying the solar cell model used. That is, it was desired to incorporate arbitrary solar cell characteristic I-V curves, temperature coefficients, and cell resistances. The method of expressing cell characteristics in closed equation form was abandoned, and point-by-point descriptors were substituted.

4.1 SOLAR CELL MODEL

The previously developed computer program used an equation of the form

$$I = \alpha - \beta e^{\gamma V} \quad (\text{where } I = \text{current, } V = \text{voltage})$$

as a model for the solar cell I-V curve. The constants α , β , and γ were manually computed for a given cell, temperature, and solar intensity. At a different temperature the origin of coordinates was assumed to undergo a translation proportional to the temperature change, ΔT . The translation equations were of the form

$$\Delta I = -T_c (\Delta T) ;$$

$$\Delta V = T_v (\Delta T) + (\Delta I) R_s ,$$

where T_c , the short-circuit temperature coefficient, and T_v , the open-circuit voltage coefficient, were each expressed as piecewise linear functions of solar intensity. The cell resistance is R_s .

Thus, for an arbitrary temperature, the current was

$$I = \alpha + \Delta I - \beta e^{\gamma(V-\Delta V)},$$

while the power became

$$p = IV = V [\alpha + \Delta I - \beta e^{\gamma(V-\Delta V)}].$$

Using the Newton-Raphson method, the maximum power point could be easily calculated from the above power equation.

Finally, for solar intensities differing from the initially specified intensity, the solar cell power was assumed to be a linear function of intensity.

The present computer program does not use any specific equation to represent the initial I-V curve. Instead, the curve is defined by reading into the computer up to 200 I-V points. Linear interpolations and extrapolations are then used to evaluate the curve at unspecified points.

The linear shift of the curve with change in temperature, which was assumed in the previous program, is still incorporated in the present version. However the calculation of T_c and T_v is now performed by inputting up to 25 values of each quantity and then linearly interpolating.

Up to this point the only differences between the previous program and the present program were implementational (i.e., instead of an equation for the I-V curve, data points were read in, etc.). Now, the first procedural difference: Instead of linearly scaling the maximum power of a cell (with solar intensity), the vertical axis of the I-V curve is linearly shifted proportional to intensity. (Thus the short-circuit current is assumed to vary linearly with intensity.) With the input

data points shifted, the maximum power point is calculated by means of a binary search for the maximum of the expression $P = IV$, where P is the power output of the cell.

Finally, the open-circuit voltage is computed from the intersection of the abscissa with the input data points. This procedure results in a solar cell model whose short-circuit current is proportional to light intensity, at constant temperature, and whose open-circuit voltage changes logarithmically with intensity, provided a curve is specified having the normal exponential solar cell I-V curve. Temperature changes cause the origin of coordinates to undergo an effective translation along a curve whose slope depends upon the temperature coefficients used. The temperature coefficients may be arbitrary functions of solar intensity. All data points for the solar cell room temperature curve and the temperature coefficients may be spaced nonuniformly for input.

4.2 INPUT/OUTPUT LISTS

Several data cards are read by the program. The following subsections describe their order, contents, and format.

4.2.1 INPUT CARDS

The first input card specifies the cell resistance, R_s . It may be punched anywhere in the first ten columns, with a decimal point.

The second card contains the first value of the short-circuit temperature coefficient, T_c , and the value of the solar intensity associated with it. The units are milliamps $^{\circ}\text{C}^{-1}$ and milliwatts cm^{-2} . The next card contains the second value, etc. Up to 25 points may be thus specified. The values of T_c and brightness are read from anywhere in columns 1 to 10 and 11 to 20, respectively. The final card of this set must have a 7 punched in column 80 as a delimiter for this data set.

The next set of cards defines values of the open-circuit voltage temperature coefficient, T_v , as a function of solar intensity. Units are millivolts $^{\circ}\text{C}^{-1}$ and milliwatts cm^{-2} . The input format is identical to that of T_c , including the 7 punched in column 80 of the last card of the set.

Finally, the set of points (up to 200) defining the I-V curve is read. The current (in mA) is punched anywhere in columns 1 through 10, while the corresponding voltage (in volts) is punched anywhere in columns 11 through 20. Again, the final card of the set must contain a 7 punched in column 80.

In addition to the previously mentioned inputs, three additional cards are required for each run and must be read in order:

Card 1 - Title Card

Column 1 - Must contain a 1, which identifies it as the title card.

Columns 2 to 71 - Can be filled with any alphanumeric characters to identify the run.

Columns 72 to 80 - Not used

Card 2 - Input Data

Column 1 - Must contain a 2, which identifies it as the data card.

Columns 5 to 12 - Contains date the run is to start. The format is Mo/Da/Yr. (earth time)

Columns 15 to 22 - Contains date run is to be terminated. Same format as above.

Columns 25 to 32 - Contains date payload landed on surface of Mars. Same format as above.

Column 34 - Not used.

Column 35 - Cloud model flag. Presently not implemented.

Column 37 - Dust storm flag: 1, if storm is in progress; 0, if no storm is in progress.

NOTE: When "no storm" is specified, the program assumes that a storm has just ended and dust particles are just starting to fall on the panel.

Column 39 - Radiation model flag. If 1, use no attenuation model. If 2, use worst-case attenuation model (21%/year).

Columns 40 to 44 - Latitude of solar panel. Zero is at the equator; $+90^{\circ}$ is north pole; -90° is south pole.

Columns 45 to 48 - Angle of solar panel relative to the ground; 90° is a vertical panel.

Columns 50 to 54 - This is the azimuth angle of the panel defined as follows: Let a vector normal to the panel (pointing out) be projected on the local horizontal plane. The direction of this projection measured clockwise from north is the azimuth angle. Thus, for a panel tilted so as to receive the morning sun at any point above the eastern horizon, the azimuth angle would be 90° .

Columns 56 to 60 - Active area of solar panel in square feet.

Columns 61 to 65 - Desired power output in watts. For each instantaneous power point, the program calculates and prints the panel area required to generate this power.

Columns 66 to 70 - Defines the horizon below which the sun will not be able to shine on the solar panel. Can be used to indicate the blockage of sunlight, e.g., by a hill that rises 15° above the horizon.

Columns 72 to 74 - Days print interval. If 1, prints daily summary every day; if 3, prints daily summary every third day, etc.

Columns 76 to 78 - Daily frequency print interval. Specifies the number of times within any Mars day (that a summary is printed) to print the instantaneous value of parameters. If 0, no instantaneous power points are printed (only the summary is printed). If 1, only the maximum power point is printed. If 2 or greater, the points in time for dawn, sun on panel, maximum power, sun off panel, and dusk are printed. In addition, this number is divided into the period of one Mars revolution and, starting from midnight, instantaneous power points are printed at this interval.

Card 3 - Input Data (continued)

Column 1 - Must contain a 3, which identifies it as the last data card.

Columns 2 to 10 - Value of ρ , the ratio of dust particle circumference to light wavelength. It can range from 2π to 200π .

Columns 11 to 20 - Value of N , the index of refraction of the dust particle. It can range from 1.4 to 1.8.

Columns 21 to 30 - Value of n , the number density of the dust particles in the atmosphere. It can range from 10^{15} to 10^{17} km^{-3} . (E format with the E to be punched in column 27)

Columns 31 to 40 - Value of a , the radius of the dust particles. It can range from 5×10^{-10} to $5 \times 10^{-8} \text{ km}$. (E format with the E to be punched in column 37)

Columns 41 to 50 - Value of ζ , the absorption coefficient of the dust. It can range from 10^5 to 10^7 km^{-1} . (E format with the E to be punched in column 47)

Columns 51 to 60 - Value of ν , the reflection coefficient of the dust layer. It ranges from 0 to 1.

4.2.2 OUTPUT DATA

Column 1 - Time of Martian day in hours past midnight. (Martian midnight ≈ 0.0000 .)

Column 2 - The power output of one 4 cm^2 solar cell in milliwatts.

Column 3 - The power output, in watts, of a panel of 4 cm^2 solar cells whose area is specified in ft^2 as an input parameter.

Column 4 - The panel area required, in ft^2 , to provide a power output specified as an input parameter.

Column 5 - The accumulated energy, in watt-hours, generated by the panel since the start of the computer run.

Column 6 - Solar intensity at the upper atmosphere of Mars in milliwatts/cm².

Columns 7, 8, 9 - Solar intensity transmission coefficients for the Mars atmosphere, cloud cover, and dust environment, respectively.

Column 10 - Magnitude of solar intensity, in mW/cm², after passing through the Mars atmosphere, cloud covers, and dust environment.

Column 11 - Solar intensity normal to panel surface in mW/cm².

Column 12 - Temperature of solar panel in °C.

Column 13 - Power output, in mW, of 4 cm² solar cell before radiation degradation is applied.

Column 14 - Percentage of column 13 actually generated due to radiation degradation.

Column 15 - Elevation of sun in degrees as measured from the local geographic vertical.

Column 16 - Azimuth of sun in degrees. That is, the angle measured in the local geographic horizontal plane between the projection of the sun-Mars line in that plane and Mars North (measured in the clockwise direction).

Column 17 - Elevation of the sun relative to the solar panel surface vertical.

Line preceding detail printout:

"Mars day" - The number of Martian days that have elapsed since start of computer program.

"Earth time" - The number of Earth days that have elapsed since midnight (GMT) of April 20, 1967 to midnight of this Mars day. Adding column 1 time to this time gives Earth GMT of that particular point.

Printed at the bottom of the page are the quantities:

- (a) Short-circuit current (in mA),
- (b) Maximum power point voltage (in volts), and
- (c) Open circuit voltage (in volts),

for the three periods of the day:

- (a) Near dawn,
- (b) At the time of maximum power, and
- (c) Near dusk.

The sun elevation angle for the "near dawn" and "near dusk" periods is also presented.

4.3 PROGRAM LISTING

The following is a literal copy of the cards comprising the computer program. Language used is Fortran IV, and no compilation or other messages are included. These data cards were used to run the sample problem.

RJOR 5411-RISHOP,F

RASSTGN M951,1)DEVICE,SI*,1)FBCD*

RASSTGN F960,1)DEVICE,SI*,1)FBCD*

RASSTGN F961,1)DEVICE,LO*

RFORTRAN L5,GO,NS,ADP

REAL MAAEP,MOECC,MICRO,LAT,LOSS

INTEGER AIFG,CLDFG,DSIFG,CARID,CFG,DFG,ELEFG

INTEGER RADFG,DSI,1PI,OUT,SMO,SDAY,SYR,EMO,EDAY,EYR

INTEGER TITLE, START, END

COMMON SMO,SDAY,SYR,EMO,EDAY,EYR,ALMO,LDAY,LYR,AIFG,CLDFG,DSIFG

COMMON RADFG,LAT,CLCPE,ALZINH,AREA,PUOI,HOKI2,DSI,1PI,DELPT

COMMON MICR,IN,TRUEV,SEWIN,CUGIN,DELDY,TITLE,ASOMU,LPAGE,RAVLO

COMMON OUT,IMPFU,DELLI,MAEP,EUPAK,MOECC,EPJL,ERATI,EARIB,SEMIM

COMMON START,END,LANDI,LOSS

COMMON CURENT (200), VOLTAG (200), NIVPTS, JCLEMP (25), DSCI (25),

* NSCPTS, JCLEMP (25), DUCT (25), NOCPTS, RESJCT

COMMON /BERNIE/ BNRHU, BNENN, BKSIN

COMMON /DUSTY/ PARAMA, PARAHIL, PARAMV

DIMENSION TITLE (32)

IN = 60

OUT = 61

READ (IN, 2) RESJCT

NSCPTS = 0

11 NSCPTS = NSCPTS + 1

READ (IN, 2) JCLEMP (NSCPTS), DUCT (NSCPTS), NO MORE

IF (NO MORE) 12, 11, 12

12 NOCPTS = 0

13 NOCPTS = NOCPTS + 1

READ (IN, 2) JCLEMP (NOCPTS), DUCT (NOCPTS), NO MORE

IF (NO MORE) 4, 13, 4

4 NIVPTS = 0

3 NIVPTS = NIVPTS + 1

READ (IN, 2) CURENT (NIVPTS), VOLTAG (NIVPTS), NO MORE

2 FORMAT (2F10.0,59X11)

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

C * * * * *

7254LL4

7254CON1

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

7254CON0

```

CEGIN = COS(EGINC)
DELDY=1.027486426
9 READ(IN,8) CARID,TITLE
8 FORMAT(I1,35A2)
IF(CARID-1)7,6,7
7 WRITE(OUT,21)
21 FORMAT(40H)TITLE CARD NOT WHERE IT SHOULD BE - JOB ADJUSTED)
GO TO 9
C READ SET-UP INFORMATION
6 READ(IN,10) CARID,SMU,SDAY,SYK,EMO,EDAY,EYR,LNU,LDAY,LYR,ATFG,CLDF,7254ELLE1
1G,CFG,DJIFG,DFO,RADFG,LAT,SLOPE,AZIMH,AREA,POUT,ELEFG,HORIZ,DSI, 7254ELLE1
21P1
10 FORMAT(I1,1X,3(2X,12,1X,12,1X,12),1X,6(11,F5,1,F4,1,1X,F5,1,1X,
IF5,2,F5,1,1,F5,1,1X,13,2X,13)
AGOM1=0.0
IF(CARID - 2)5,13,5
5 WRITE(OUT,20)
20 FORMAT(50H)INPUT PARAMETER CARD NOT WHERE IT SHOULD BE - JOB ADJUSTED)
*TED)
GO TO 9
15 READ (IN, 68) CARID,BKSNH, UNENH, BKSN, PARAMH, PARAMV
68 FORMAT(I1,F5,0,F10,C,3E10,2,F10,0)
IF(CARID - 3) 69, 67, 69
69 WRITE (OUT, 70)
70 FORMAT (33H)INPUT CARD MISSING - JOB ADJUSTED)
GO TO 9
67 LPAGE = 1
14 FORMAT(I1,20X,72H) A K 3 3 U L A R P 6 W E R A N A L Y S I S 7254ELLE1
1P R O G R A M REV 012X4HPAGE14725X12HROW TITLE...55A2)
IF(PI1)40,50,40
50 DELPT=1.028
GO TO 41
40 DELPT=DELDY/1P1
41 WRITE(OUT,14)LPAGE,TITLE
WRITE(OUT,16)SMU,SDAY,SYK,EMO,EDAY,LYR,LNU,LDAY,LYR
16 FORMAT (21H)ANALYSIS PERIOD FROM13,IN/12,1H/12,GN TO 12,1H/1
*2,1H/12/25H SPACECRAFT LANDING DATE 12,1H/12,1H/12//26H ATTENUATI
*ON FACTORS USED-
WRITE (OUT, 17) PARAMH, UNENH, UNENH, BKSN, PARAMH, PARAMV
17 FORMAT (1/2 SMALL A =E10,2E NM/2 NMU =E10,2E LARG N =E10,2E NM
*ALL N =E10,2E NM(-3)E/2 ZELA =E10,2E NM(-11)E/2 NU =E10,3)
WRITE (OUT,18) RADFG
18 FORMAT (1/2 NAVIGATION MODEL FLAG =E12)
IF (DSIFG) 25, 24, 25
25 WRITE (OUT, 19)
19 FORMAT (1/2 DUST STORM IN PROGRESS)
GO TO 32
24 WRITE (OUT, 28)
28 FORMAT (1/2 NO DUST STORM)
32 WRITE(OUT,37)LAT,SLOPE,AZIMH,AREA,POUT,HORIZ,DSI 7254ELLE1
37 FORMAT (1/21H SPACECRAFT POSITION-71 H LATITUDE20.1,4H DEG/ 21H
* PANEL INCLINED FROM/10H HORIZONTAL BYF14.1,4H DEG/22H DIRECTIO
*N OF INCLINE/12H (AZIMUTH)F18.1,4H DEG//17H SOLAR PANEL AREA13.

```

*1.7H SW.FT./22H REQUIRED POWER OUTPUT F8.1.5HWATTS//16H DEFINED HOR

*12UMF14.1,4M DEG//17H PRINTER CONTROL-72H PRINT(SUMMARY EVERY14

7254ELE2

7254ELE1

7254ELE1

IF(IPI-3)53,39,39

53 IF(IPI)42,43,42

43 WRITE(OUT,44)

44 FORMAT (22H NO DETAIL PRINT-OUT)

7254ELE1

7254ELE1

7254ELE1

GO TO 52

42 IF(IPI-1)46,47,46

47 WRITE(OUT,48)

48 FORMAT (24H PRINT MAXIMUM POWER DETAIL ONLY)

7254ELE1

7254ELE1

7254ELE1

GO TO 49

46 WRITE(OUT,51)

51 FORMAT (44H PRINT MAX. POWER, DAWN, DUSK DETAILS ONLY)

7254ELE1

7254ELE1

7254ELE1

39 ELEVS=DELPT*24.

WRITE(OUT,56) ELEVS

56 FORMAT (22H PRINT DETAIL EVERY F8.3,20H HRS DURING DAYLIGHT)

7254ELE1

7254ELE1

7254ELE1

49 WRITE(OUT,57)

57 FORMAT (20H (ONLY ON SUMMARY DAYS))

7254ELE1

7254ELE1

7254ELE1

52 HOR12=HOR12/RADEG

DEGRA = (1. - LOGS)*100. + .005

WRITE(OUT,58)DEGRA

01 FORMAT(102H SOLAR PANEL OUTPUT BASED ON MAKINER 1969 S/N 2, 2X2 C

*M SOLAR CONVERTER I-V CURVE, H/P .018 IN, 2 OHM-CM/28H MISCELLANEOU

*S POWER LOSSES F5.2,9H PER CENT)

SYR=SYR+1900

EYR=EYR+1900

LYR=LYR+1900

CALL JTIME(SMO,SDAY,SYR,START)

CALL JTIME(EMO,EDAY,EYR,END)

CALL JTIME(LMO,LDAY,LYR,LANDT)

SYR=SYR-1900

CALL PART B

GO TO 1

END

RFORTRAN L5,60,INS,ADP

SUBROUTINE PART B

C	TOTAL ENERGY GENERATED SINCE START OF RUN	7254DEF2
C	INSTANTANEOUS PRINT FREQUENCY 124/1PI== OF TIMES/DAY*	7254DEF2
C	ALLOWABLE TIME ERROR IN LOCATING MAXIMUM POWER	7254DEF2
C	POINT IN DAYS 1.0000E-1 MINUTE*	7254DEF2
C	CONVERSION FACTOR FROM RADIANS TO DEGREES	7254DEF2
C	LOGICAL UNIT NUMBER OF DEVICE INPUT DATA IS ON	7254DEF2
C	LOGICAL UNIT NUMBER OF DEVICE RESULTS ARE TO BE	7254DEF2
C	PRINTED ON	7254DEF2
C	LENGTH OF MARTIAN DAY RELATIVE TO EARTH DAY	7254DEF2
C	CONTAINS NUMBER THAT IDENTIFIES THE TYPE OF INPUT	7254DEF2
C	RECORD	7254DEF2
C	CONTAINS ALPHANUMERIC CHARACTERS FROM INPUT RECORD	7254DEF2
C	NUMBER 1	7254DEF2
C	MONTH NUMBER OF PROBLEM END TIME	7254DEF2
C	DAY OF PROBLEM END TIME	7254DEF2
C	YEAR OF PROBLEM END TIME	7254DEF2
C	MONTH NUMBER OF PAYLOAD LANDING DATE	7254DEF2
C	DAY OF PAYLOAD LANDING DATE	7254DEF2
C	YEAR OF PAYLOAD LANDING DATE	7254DEF2
C	CLOUD ATTENUATION MODEL TO USE - NOT IMPLEMENTED	7254DEF2
C	NOT USED	7254DEF2
C	DUST ATTENUATION MODEL TO USE - NOT IMPLEMENTED	7254DEF2
C	NOT USED	7254DEF2
C	RADIATION ATTENUATION MODEL TO USE IF 0 AND ATTEN	7254DEF2
C	1, 21/YEAR	7254DEF2
C	AREA OF SOLAR PANEL IN SQ. FT.	7254DEF2
C	INPUT PARAMETER - DESIRED POWER OUTPUT IN WATTS	7254DEF2
C	IF 0 USE REFERENCE I-V CURVE FOR SOLAR CELL	7254DEF2
C	1 READ I-V PARAMETERS OFF OF NEXT INPUT RECORD	7254DEF2
C	DAILY SUMMARY PRINTOUT INTERVAL	7254DEF2
C	OUTPUT PAGE COUNTER	7254DEF2
C	INSTANTANEOUS RESULTS PRINT INTERVAL IN MARTIAN DAYS	7254DEF2
C	NEXT POINT IN TIME TO OUTPUT INSTANTANEOUS RESULTS	7254DEF2
C	PREVIOUS VALUE OF UNIT CELL POWER	7254DEF2
C	LANDING DATE IN DAYS FROM APRIL 20, 1967	7254DEF2
C	MIDNIGHT OF CURRENT DAY	7254DEF2
C	IF 1, PRINT SUMMARY AT END OF CURRENT DAY	7254DEF2
C	COUNT BETWEEN DAYS SUMMARY IS TO BE PRINTED	7254DEF2
C	IF 0, BYPASS ADVANCING IN TIME IN SLITTING UP FOR NX	7254DEF2
C	NUMBER OF MARTIAN DAYS SINCE START DAY	7254DEF2
C	PRINT	7254DEF2
C	POUNE	7254DEF2
C	POWFG	7254DEF2
C	LINEC	7254DEF2
C	AGUIN - = OF LINES PRINTED ON CURRENT OUTPUT PAGE	7254DEF2
C	AGUIN - ANGLE OF INCIDENCE OF SUN ON PANEL	7254DEF0
C	AGUIN - MOST PREVIOUS VALUE OF AGUIN	7254DEF0
C	ATI - ATTENUATION DUE TO ATMOSPHERE	7254DEF0
C	ATI - IF 0 USE NOMINAL MODEL IF 1, USE WORST CASE	7254DEF0
C	ATMAT - SUB-ROUTINE THAT CALCULATES ATMOSPHERIC ATTENUATION	7254DEF0

C	AZIMH	-	AZIMUTH ANGLE OF FALL LINE OF HILL S/C SITS ON	72540DEF0
C	BRITC	-	SOLAR INTENSITY AT MARKS UPPER ATMOSPHERE	72540DEF0
C	CAZHL	-	COSINE OF AZIMH	72540DEF0
C	CEWIN	-	COSINE OF EWING	72540DEF0
C	CLAT	-	COSINE OF LAT	72540DEF0
C	CLONG	-	COSINE OF LONG	72540DEF0
C	CPMER	-	COSINE OF PMERU	72540DEF0
C	CSLOP	-	COSINE OF SLOPE	72540DEF0
C	CTRUE	-	COSINE OF TRUEI	72540DEF0
C	DAWN	-	TIME OF SUNRISE IN EARTH TIME	72540DEF0
C	DAMNF	-	IF 1, SUN AT DAWN	72540DEF0
C	DAYFO	-	IF 1, SUN ABOVE HORIZ. IF 0, SUN BELOW HORIZ.	72540DEF0
C	DAYWH	-	ENERGY GENERATED BY SOLAR ARRAY FOR CURRENT MARKS DAY	72540DEF0
C	DELT1	-	CALCULATION INTERVAL (IN EARTH DAYS AND FRACTION OF DAYS)	72540DEF0
C	DUSK	-	TIME OF SUNSET IN EARTH TIME	72540DEF0
C	DUSNF	-	IF 1, SUN AT DUSK	72540DEF0
C	EARTS	-	SOLAR INTENSITY AT 1 ASTRONOMICAL UNIT (AU)	72540DEF0
C	EDPMR	-	NUMBER OF EARTH DAYS PER MARS REVOLUTION	72540DEF0
C	EK1	-	TRIAL VALUE OF ECCENTRIC ANOMALY	72540DEF0
C	EK2	-	NEW TRIAL VALUE OF ECCENTRIC ANOMALY	72540DEF0
C	ELEMI	-	PREVIOUS VALUE OF ELEVS	72540DEF0
C	ELEVS	-	SUN ELEVATION ANGLE AS MEASURED FROM LOCAL HORIZON	72540DEF0
C	EPSIL	-	ALLOWABLE TRUNCATION ERROR IN CALCULATING ECCENTRIC ANOMALY	72540DEF0
C	EWING	-	INCLINATION OF EQUATORIAL PLANE TO ORBIT	72540DEF0
C	ERATI	-	RATIO USED TO CALCULATE TRUEI (SEE CONSTANTS LIST)	72540DEF0
C	HORIZ	-	HORIZON ANGLE. ELEVATION ANGLE OF DAY-NIGHT LINE	72540DEF0
C	JTIME	-	SUBROUTINE TO CONVERT FROM DAY-MONTH-YEAR TIME	72540DEF0
C	LAT	-	FORMAT TO NUMBER OF DAYS FROM A REFERENCE DATE	72540DEF0
C	LITZF	-	MARS LATITUDE OF SPACECRAFT	72540DEF0
C	LONG	-	IF 1, SUN ABOVE PANEL. IF 0, SUN BELOW PANEL	72540DEF0
C	MAAEP	-	LONGITUDE, NUMBER OF DEGREES FROM MERIDIAN THAT IS AT MIDNIGHT AT START TIME	72540DEF1
C	MEANI	-	MEAN ANOMALY OF MARS AT EPOCH (APRIL 20, 1967)	72540DEF0
C	MEANI	-	CURRENT MEAN ANOMALY	72540DEF0
C	MEUCC	-	MARS ORBITAL ECCENTRICITY	72540DEF0
C	MAPUC	-	MAXIMUM POWER OUTPUT OF UNIT CELL DURING DAY	72540DEF0
C	OFFPU	-	IF 1, SUN JUST OFF SOLAR PANEL	72540DEF0
C	UNFG	-	IF 1, SUN JUST ON SOLAR PANEL	72540DEF0
C	PASS1	-	FLAG TO INDICATE FIRST PASS THROUGH CALCULATIONS	72540DEF0
C	PMERU	-	ANGLE BETWEEN VERNAL EQUINOX AND PRIME MERIDIAN	72540DEF0
C	SAZHL	-	SINE OF AZIMH	72540DEF0
C	SDAY	-	DAY OF PROBLEM START TIME	72540DEF0
C	SEMIH	-	SEMI-MAJOR AXIS OF MARS ORBIT	72540DEF0
C	SEWIN	-	SINE OF EWING	72540DEF0
C	SLAT	-	SINE OF LAT	72540DEF0
C	SLONG	-	SINE OF LONG	72540DEF0
C	SLOPE	-	VERTICAL SLOPE ANGLE OF HILL S/C SITS ON	72540DEF0
C	TMU	-	MONTH NUMBER OF PROBLEM START TIME	72540DEF0
C	TPMER	-	SINE OF PMERU	72540DEF0
C	SLOP	-	SINE OF SLOPE	72540DEF0
C	START	-	TI AT START OF PROBLEM	72540DEF0

C	STRUE	-	SINE OF TRUEI	7254DEFO
C	SYR	-	YEAR OF PROBLEM START TIME	7254DEFO
C	TI	-	CURRENT TIME (IN EARTH DAYS AND FRACTION OF DAYS)	7254DEFO
C	THOLD	-	TEMPORARY STORAGE FOR T1 WHILE SPECIAL POINTS ARE CALCULATED (SUCH AS DAWN)	7254DEFO
C	TRUEI	-	CURRENT TRUE ANOMALY	7254DEFO
C	TRUEV	-	TRUE ANOMALY OF MARKS VERNAL EQUINOX	7254DEFO
C	AX	-	SUN X COMPONENT VECTOR IN LOCAL GEOCENTRIC FRAME	7254DEFO
C	AY	-	SUN Y COMPONENT VECTOR IN LOCAL GEOCENTRIC FRAME	7254DEFO
C	YHILL	-	SUN Y COMPONENT VECTOR IN LOCAL GEOCENTRIC FRAME	7254DEFO
C	ZU	-	SUN Z COMPONENT VECTOR IN LOCAL GEOCENTRIC FRAME	7254DEFO
C	ZHILL	-	SUN Z COMPONENT VECTOR IN LOCAL GEOCENTRIC FRAME	7254DEFO
C	REAL NAMEP	-	NAMEP	
C	REAL	-	NAMEP	
C	REAL LAT, LOSS	-	NAMEP	
C	INTEGER ATFG, CLDFG, DUSTFG	-	NAMEP	
C	INTEGER DAWN, RADFG	-	NAMEP	
C	INTEGER POWFG, PDONE, DUCK, DAWN, ONFG, OFFFG, LITEF, PRTOF, DPTCT, OSI	-	NAMEP	
C	INTEGER IPI, PRINT, OUT	-	NAMEP	
C	INTEGER PASSI	-	NAMEP	
C	INTEGER JMO, DAY, YR, EMU, DAY, LYR, START, END	-	NAMEP	
C	INTEGER DAYFG, DAYCT	-	NAMEP	
C	INTEGER TITLE	-	NAMEP	
C	COMMON JMO, DAY, YR, EMU, DAY, LYR, LNO, LDAY, LYR, ATFG, CLDFG, DUSTFG	-	NAMEP	
C	COMMON RADFG, LAT, SLOP, AZI, H, RADEG, POOT, HORIZ, OSI, IPI, DULPT	-	NAMEP	
C	COMMON MICRO, IN, IRUEV, SEMIN, CEGIN, DELDY, TITLE, AGGUMU, EPAGE, RADEG	-	NAMEP	
C	COMMON OUT, IMPFG, DELT, IN, RADEG, CUP, IN, RADEG, EPSIL, GERATI, LANTO, SEMIN	-	NAMEP	
C	COMMON JAKI, LEND, LANDI, LOSS	-	NAMEP	
C	COMMON CRUO (400), KRUD, BARK (50), NURD, TRASH (50), JUNK, GUOF	-	NAMEP	
C	DIMENSION M (17), TITLE (35)	-	NAMEP	
C	DIMENSION SINGNG (2), JCCEND (3), VMPEND (3), UCVENO (3)	-	NAMEP	
C	14 FORMAT (I1, 20X, 72H A K 3 5 U L A R P U W E R A N A L Y S I S 7254ELE1	-	NAMEP	
C	IP R O G K A M REV 012X4HPAGE14725X12HRUN TITLE...35A2)	-	NAMEP	
C	NOL=45	-	NAMEP	
C	HOS=24	-	NAMEP	
C	PJCH=0	-	NAMEP	
C	DAYWH=0	-	NAMEP	
C	TOTWH=0	-	NAMEP	
C	CSLOP=COS(SLOP/RADEG)	-	NAMEP	
C	SSLOP=SIN(SLOP/RADEG)	-	NAMEP	
C	CAZHL=COS(AZI/H/RADEG)	-	NAMEP	
C	SAZHL=SIN(AZI/H/RADEG)	-	NAMEP	
C	PTIME=JAKI+DELPT	-	NAMEP	
C	ENDAT=JAKI+DELDY	-	NAMEP	
C	PRIOT=1	-	NAMEP	
C	DPTCT=DSI	-	NAMEP	
C	KUFG=0	-	NAMEP	
C	DAYCT=1	-	NAMEP	
C	TOTWH=0	-	NAMEP	
C	PRINT=0	-	NAMEP	
C	PDONE=0	-	NAMEP	
C	POWFG=0	-	NAMEP	

DUSKF=0	7254EL
DAWNF=0	7254EL
ONFG=0	7254EL
OFFEG=0	7254EL

LITEF = 0	7294ELEB2
DAYFG = 0	72541CNO
FLEW1 = 0.0	72541CNO
LONG = 0.0	72541CNO
SLONG = SIN(LONG)	
CLONG = COS(LONG)	
LONG = 1	

```

PROG=
SLAT = SIN(LAI/ 57.29577951)
CLAT = COS(LAI/ 57.29577951)
PI = 3.141592653589793
KSTAGE = 1

```

GO TO 1022
180 CALL JOIN UP (TI, PIERD, ELEVS,
* EN1, EN2, EPSIL, TRUEL, PASSL, XG, YG, ZG, SLAT, CLONG, CLAT,
* # SCALING, CHAEP, COPPER, PROCC, ERATI, TRDEV, CEGIN, BEGIN)

C	IF(POWFG-1)60,310,60 TEST ELEVATION ANGLE FOR BEING ABOVE DEFINED HORIZON	7254ELE0
C	60 IF(ELEVS - HORIZ)205,275,275 JOIN BELOW HORIZON - IS IT ALMOST DAWN	7254ELE2 7254ELE1
C	265 IF(DAWN - 1)300,310,300 IT IS NOT NEAR DAWN, DID WE JUST PAST DUSK	7254ELE1 7254ELE1
C	300 IF(DAYFG - 1)320,330,320 DUSK, DID NOT PAST DUSK IS THIS DUSK	7254ELE1 7254ELE1

C	320 IF(DUONF-1)310,310,310 YES,DUON WAS PAIDLO. FIND TIME OF DUON AND RE-CALCULATE THERE 7254LE1 7254LLE2
	330 IF(ONFG-1)331,310,331

```

DAYFG = 0
DOURN = 1 - DEL(I) + LLEMI*DEL(I)*(ELEMT - ELEVS)
DOURN = 1
HOLDL= PPERD
PPERD = PPERD + 6.124211Q8*(DOURN - TI)
THOLD = TI
TI = DOURN

```

60 TO 180
IN ABOVE MORTON - IS IT ABOUT AT DUSK
7254ELE1
7254ELE1

275 IF(DUNT - 11350,310,350
 NOT AT DUNN. DID WE JUST PASS THRU DAWN
 7234ELE1

350 IF (DAYFQ) 310,350,310	7254ELE
JUST PASSED MKU DAWN. FIND TIME OF DAWN AND RE-CALCULATE	7254ELE
360 DAWN = 1 - ELEM1 + ELEMDELTA / (ELEM1 - ELEV3)	7254ELE

HOLD = PMERD	7254ELE1
PMERD = PMERD + 6.12421108*(DAWN - 11)	7254ELE1
HOLD = 11	7254ELE1

TI=Dawn
SAGE - 1
7454ELE1

MAXPW=0.
 DAYNH = 0.0
 DAYFG = 1
 GO TO 160

310 CALL LNC AUG (XG, YG, ZG, SOLUP, CSLUP, CAZHL, SAZHL, AGUIN)

C IF SUN ABOVE OR BELOW PANEL
IF(AGUIN) 370,390,380 7254ELE1
7254ELE1

C IF AT DAWN, GO PRINT II
C IF AT DAWN + SUN ALMOST ON PANEL, ASSUME IT IS ON PANEL

C 370 IF(DAWN-1)371,372,371 7254ELE2
372 IF(AGUIN-ELEV*(0.02)340,430,430 7254ELE1
7254ELE1

C 371 IF(ONPG-1)390,390,390 7254ELE1
C NOT JUST ON, DID SUN JUST GO OFF OF PANEL

C 390 IF(LITEF - 1)400,410,400 7254ELE1
C IS THIS THE CALCULATION OF THE OFF POINT

C 400 IF(OFFG - 1)401,340,401 7254ELE3
401 IF(ONPG-1)400,340,400 7254ELE2
7254ELE1

C 410 OFFG=1
LITEF = 0
IF(DUCKF-1)411,340,411 7254ELE1

C 411 THOLD = II
HOLD=P.MERU
II = II - DELTI + AGUIN*DELTII/(AGUIN AGUIN) 7254ELE1

C P.MERU = P.MERU + 0.12421108*(II-THOLD) 7254ELE1
GO TO 180 7254ELE1

C 380 IF(DUCKF-1)381,410,381 7254ELE1
C 381 IF(OFFG-1)420,340,420 7254ELE1

C 420 IF(LITEF - 1)420,340,420 7254ELE1
C SUN IS ON PANEL, DID IT JUST GET ON

C 430 ONPG = I
LITEF = I 7254ELE1

C 440 THOLD = II
HOLD = P.MERU 7254ELE1

C 340 BRITE = EARLY/(SUMIN*(1.0 - DECCOS(EN2)))**2
C CALCULATE ATTENUATION OF SUN BETWEEN UPPER ATMOSPHERE AND
PANEL FACE. 7254ELE1

C CALL ATMAT(ELEV,AT10,AT11)
CALL CLDPA(THUL,PLA,CLDPA,AL2) 7254ELE1

C KUNYM = (II - START) * 0.6400*0
CALL DJAT (AGUIN, KUNYM, DJTFG, AT3) 7254ELE1

C SUNRY = UNITE * AT1 * AT2 * AT3
SNPOW = SUNRY * SIN (AGUIN) 7254ELE1

C CALCULATE JOLAR PANEL TEMPERATURE
IF(ONPG-1)450,460,460 7254ELE1

C PHI=90./RADLUT*AGUIN
IF(SNPOW)450,460,460 7254ELE1

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450 SINDW = 0.0 7254LELE1
460 AINDG=SINDW 7254LELE2
CALL TEMP(LAT,IFDWN,AINDU,PHI,TAPFG,PTEMP)

KALGLD = 0
GO TO (13, 17, 98), KSTAGE
15 IF (ELEVJ) 98, 98, 20
25 IF (ELEVJ - 0.262) 98, 19, 19
19 KALGLD = 1
CALL POWER (PTEMP, SINDW, KAPW, VMPEND (1), SCCEND (1), OCVEUD
* (1), KALGLD)
SUNANG (1) = ELEVS * 57.29578
KSTAGE = 2
GO TO 99

17 IF (ELEVJ - 0.262) 21, 21, 98
21 KALGLD = 1
CALL POWER (PTEMP, SINDW, KAPW, VMPEND (3), SCCEND (3),
* OCVEUD (3), KALGLD)
SUNANG (2) = ELEVS * 57.29578
KSTAGE = 3
GO TO 99

98 CALL POWER (PTEMP, SINDW, KAPW, VRAPW, SCURNT, OCVEUD, KALGLD)
99 CALL RADATT (TEMP, SINDW, KAPW, VRAPW, SCURNT, OCVEUD, KALGLD)
POWUC = RAPW * AT4 * LOSS

IF (CONFG = 1) 470, 480, 490
C RECORD TIME SUN CAME ONTO PANEL

480 CONTIN = 11
470 IF (POWUC - 1) 490, 610, 490
490 IF (ELTER - 1) 511, 520, 511
520 IF (CONFG - 1) 530, 540, 530
540 DELPW = INCU - 11
545 IF (POWUC - 1) 550, 560, 550
511 IF (OFFG - 1) 570, 580, 570
570 IF (DOJNF - 1) 590, 580, 590
580 DELPW = 11 - INCU + DELTI
590 DELWH = 0.5 * DELPW * (POWUC + POWUC)

DAYWH = DAYWH + DELWH
TOTWH = TOTWH + DELWH
DELPW = DELTI
GO TO 545

560 IF (POWUC - 1) 600, 610, 600
600 IF (POWUC - POWUC) 620, 550, 550
620 POWUC = 1
KAPW = POWUC

DAYWH = DAYWH + DELWH
TOTWH = TOTWH + DELWH
PHOLD = 11
LONGH = PHOLD

650 TI = 11 - MICRO
PJCMI = POWUC
PJCMI = PHOLD - 0.12421108 * MICRO
GO TO 180

610 IF (POWUC - POWUC) 630, 640, 640
640 IF (TI - 0.11111111) 630, 650, 630

630 POONE = 1

MAXPW = 5.574*AKELA*(101*W1+5*(11-PRHOLD+DEL11))*(PUCM1+MAXPW))

IF(PRIODF - 1)660,670,660

670 IF(PIPI-1)660,700,700

550 PUCM1 = POWUC

IF(PRIODF - 1)600,690,600

690 IF(PIPI - 2)600,710,710

750 IF(DAWN - 1)720,730,720

730 WRITE(001,731)

731 FORMAT(17H *** DAWN ***)

GO TO 700

720 IF(UNFG - 1)770,740,770

740 WRITE(001,741)

741 FORMAT(20H *** JOIN UN PANEL ***)

GO TO 700

710 IF(DAWN + UNFG - 2)750,760,750

760 WRITE(001,761)

761 FORMAT(20H *** DAWN - JOIN UN PANEL)

GO TO 700

770 IF(UNFG + OFFFG - 2)780,790,780

790 WRITE(001,791)

791 FORMAT(40H *** DAWN - JOIN COL OFF OF PANEL ***)

GO TO 700

780 IF(OFFFG - 1)800,810,800

810 WRITE(001,811)

811 FORMAT(34H *** JOIN BELOW SOLAR PANEL ***)

GO TO 700

800 IF(UNSGF - 1)820,830,820

830 WRITE(001,831)

831 FORMAT(17H *** DAWN ***)

GO TO 700

820 IF(PNINT - 1)800,700,800

700 TODAY = (11 - LINDAY + DELLOY) *24.0 +0.005

APWUK = AKLA * POWUC*0.922/4.0

AS11G=KADELG*ATAN((-YG)/(-XG))

IF(-XG)840,850,850

840 AS11G=AS11G+100.

GO TO 820

850 IF(AS11G)845,855,855

845 AS11G=900.+AS11G

855 AINDG = PHIKRADELG

PHI = 90. - LLEVJ*KADELG

K1WH=101*AKELA*5.574

KWARK=(4.0*POWUC)/((POWUC*0.922)

IF(POWFG-1)856,857,856

857 M(1)=TODAY

M(2)=POWUC

M(3)=APWUK

M(4)=ALWUK

M(5) = MAXPW

M(6)=WRITE

M(7)=AT1

M(8)=AT2

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7254ELE1

POWFG = 0
NOFG=0

C IF SPECIAL CONDITION EXIST DON I UPDATE PRINT COUNTER

930 IF(NOFG)931,950,931

931 PRINT=0

IF(TI - PTIME)950,960,960

960 PTIME = TI + DELPT

PRIME = 1

950 IF(TI - ENDAY)970,980,900

980 TI = ENDAY

GO TO 990

970 IF(DUSKF - 1)910,1000,910

1000 DUSKF = 0

POWFI=0

PHERD = PHERD + 6.12421108*(ENDAY - TI)

TI = ENDAY

LITEF=0

990 ENDAY = ENDAY + DELDY

PDONE=0

DAYCT = DAYCT + 1

PTIME = TI + DELPT

OPTCT = OPTCT - 1

IF(OPTCT)1010,1020,1010

1010 PRDIF = 0

GO TO 910

1020 OPTCT = DSI

PRDIF = 1

1021 MODL=MODLPLT

IF(MODL-(MODL-LINLC))1025,1026,1022

1022 LINEC=0

LPAGE=1+LPAGE

WRITE(OUT,14)LPAGE,ITITLE

WRITE(OUT,1029) LAT

1029 FORMAT (14H LATITUDE = F3.1,4H DEG)

WRITE(OUT,1025)

1023 FORMAT(77119H TIME UNIT MAX REQUIRED ACCUM 31 AT TRANS

*ISSION 5.1. AT PANEL PANEL CELL RADIAL SUN SUN INCIDENCE

2756H OF DAY CELL PANEL AREA ENERGY UPPER FACTORS20X,40

3H TEMP POWER DEGRAD ELEV AZIM- ANGLE)

WRITE(OUT,1026)

1026 FORMAT(420H (HOUR) POWER POWER10X12H(WATT- ATMOS24X12H(MAG NORMA

*LOCALISURE FACTERGABR01H ON/9A27H(MW) (WATT)(JW.FT.) HOUR(S)9X

*47H(MW) CLOUD DUST (MW/50. CM) (CLN) RADIAT0X10H(UG) (DEG)

*PANEL/)

DPAST=TI-JART

1025 WRITE(OUT,1030)DAYCT,DPAST,PDONE,PDAYS,93X

1030 FORMAT(11H MARK DAY =,13,14H, LARTH TIME =,F9.4,23H DAYS PAST HU

*UR 0.0 OF 12,11H/12,11H/12,14H GML)

510 IF(NOFG)512,513,512

512 ELEMI=ELEVS

AGOMI=AGOMI

TI = TI + DELTI

PHERD = PHERD + 6.12421108*DELTI

7254ELE2

7254ELE2

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513 NOFG=1

1F(11-END)160,180,260

280 WRITE(OUT,281)

281 FORMAT(14H ** END OF RUN)

RETURN

END

7254ELEZ

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7254EL

RFORTRAY LS,GO,NS,ADP

SUBROUTINE ALMAT(A,LY,Z)

COMMON /BERNIE/ A, B, C

E = 1.57079632 - X

CALL NAPP(A, B, C, E, Z)

IF (Z) 1, 9, 2

1 WRITE (61, 9)

3 FORMAT (50H SET ATTENUATION EQUAL TO ZERO)

WRITE (61, 99) Z

90 FORMAT (3H ATTENUATION COEFFICIENT =E10,Z)

Z = 0.0

RETURN

2 IF (Z - 1.0) 9, 9, 4

4 WRITE (61, 9)

5 FORMAT (29H SET ATTENUATION EQUAL TO ONE)

WRITE (61, 99) Z

Z = 1.0

9 RETURN

END

7254EL

RFORTRAN L0,00,N0,0UP

SUBROUTINE SUB UP (TI, PMERU, ELEV0,

* EN1, EK2, EPSIL, IRUEI, PASSI, XG, YG, ZG, SLAT, CLONG, CLAT,

* SLONG, MAEP, LUPMK, MOECC, ERAI, IRUEV, CEGIN, SEGIN)

REAL MAEP, MEANI, MOECC

INTEGER PASSI

MEAN ANOMALY OF START DATE

180 MEANI=MAEP + (TI * 6.2031853)/EDPMK

C CALCULATE ECCENTRIC ANOMALY

EK1 = MEANI

190 EK2 = (MOECC * (SIN(EN1) - EN1 * COS(EN1)) + MEANI)/(1.0 - MOECC * COS(EN1))

*1 COSTEN1)

IF(EPSIL - ABS(EK1-EK2))200,210,210

200 EK1 = EK2

GO TO 190

C CALCULATE TRUE ANOMALY (FROM PERHELION)

210 TRUE1 = 2.0 * ATAN(EK1) * ((SIN(EK2/2.0))/COS(EK2/2.0))

C CALCULATE TRUE ANOMALY (FROM VENNAL EQUINOX)

IRUEI = IRUEI - IRUEV

C CALCULATE COMPONENTS OF MARK-JOIN VECTOR IN LOCAL GEOCENTRIC

C FRAME

IF(PASSI-1)220,230,240

230 PMERU = ATAN(CEGIN * SIN(TRUE1)/ COS(TRUE1))

IF(COS(TRUE1))240,250,250

240 PMERU = PMERU + 3.1415927

250 PASSI = 0

220 CPMER = COS(PMERU)

SPMER = SIN(PMERU)

CTRUE = COS(TRUE1)

STRUE = SIN(TRUE1)

XG=SLAT*(CLONG*(CPMER*CTRUE+SPMER*CEGIN*STRUE)+SLONG*(-SPMER*CTRUE

IE+CPMER*CEGIN*STRUE))-CLAT*SEGIN*STRUE

YG=SLONG*(CPMER*CTRUE+SPMER*CEGIN*STRUE)+CLONG*(-SPMER*CTRUE+CPME

IR*CEGIN*STRUE)

ZG=-CLAT*(CLONG*(CPMER*CTRUE+SPMER*CEGIN*STRUE)+SLONG*(-SPMER*CTRUE

IE+CPMER*CEGIN*STRUE))+SLAT*SEGIN*STRUE

ELEV0 = ATAN(ZG/ SQRT(XG**2 + YG**2))

RETURN

END

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7254ELE0

```

RFORTRAN LS,GO,NS,ADP
SUBROUTINE INC ANG (XG, YG, ZG, SSLOP, COLUP, CAZHL, SAZHL, AGOIN)
C      CALCULATE SUN ANGLE ON PANEL
310 XHILL = XG*SSLOP*CAZHL - YG*COLUP*SAZHL - ZG*SSLOP
YHILL = XG*SAZHL + YG*CAZHL
ZHILL = XG*SSLOP*CAZHL - YG*SSLOP*SAZHL + ZG*COLUP
AGOIN = ATAN(ZHILL/SURT(XHILL**2 + YHILL**2))
RETURN
END

```

7254ELE1
7254ELE1
7254ELE1
7254ELE1
7254ELE1

RFOUTRAN LS, GUIN, ADP

SUBROUTINE POWER (I, U, PL, VP, SCI, UCV, KFLAG)

IF (U) G, 8, 7

7 VMIN = 0.0

VMAX = 1.0

KF = 1

1 VP = (VMIN + VMAX)/2.0

CALL JUICE (A, VP, I, U, SCI, UI, UV, NF)

PL = A * VP

IF (VMAX - VMIN - 0.0001) 5, 5, 4

4 VPLUS = VP + 0.000025

CALL JUICE (A, VPLUS, I, U, SCI, UI, UV, NF)

PQ = A * VPLUS

IF (PQ - PL) 2, 5, 3

2 VMAX = VP

GO TO 1

3 VMIN = VP

GO TO 1

5 IF (PL) 8, 8, 10

10 IF (VP) 6, 6, 6

R PL = 0.0

VP = 0.0

SCI = 0.0

UCV = 0.0

RETURN

6 IF (KFLAG) 12, 13, 12

12 CALL UCLUCV (UI, UV, UCV)

13 IF (0.52 * U - PL) 9, 11, 11

9 SCI = SCI * 0.52 * U/PL

PL = 0.52 * B

11 RETURN

END

```

REFUKTRAN L9800,INS,ADP
SUBROUTINE JUICE (I, V, I, U, SCI, DI, DV, NFLAG)
REAL I
COMMON N/A (15), W/A (6), K/U (2), W/U (2), K/C (2), W/C (4),
* K/U (55), W/U, N/E, W/E, N/F, W/F (9), K/U (5), W/U
COMMON AMPS (200), VOLTS (200), NIV, IC (25), DIC (25), NIC,
* IV (25), DIV (25), NIV, K3

```

```

IF (KFLAG) 8, 9, 8
8 KFLAG = 0
IF (B - DIC (2)) 1, 1, 4
4 IF (B - DIC (NIC - 1)) 2, 3, 3
1 NDX = 1
GO TO 5
3 NDX = NIC - 1
GO TO 5
2 LIMIT = NIC - 1
DO 6 K = 3, LIMIT
IF (B - DIC (K)) 7, 7, 6
6 CONTINUE
STOP
7 NDX = K - 1
5 ALPHA = IC (NDX) + (B - DIC (NDX)) * ((IC (NDA + 1) - IC (NDX)) /
* (DIC (NDX + 1) - DIC (NDX)))
IF (B - DIV (2)) 11, 11, 14
14 IF (B - DIV (NIV - 1)) 12, 13, 13
11 NDX = 1
GO TO 15
13 NDX = NIV - 1
GO TO 15
12 LIMIT = NIV - 1
DO 16 K = 3, LIMIT
IF (B - DIV (K)) 17, 17, 16
16 CONTINUE
STOP
17 NDX = K - 1
15 BETA = IV (NDX) + (B - DIV (NDX)) * ((IV (NDA + 1) - IV (NDX)) /
* (DIV (NDX + 1) - DIV (NDX)))
DI = ALPHA * (1 - 26.0)
DV = (BETA * (28.0 - 7) - DI * K3) / 1.0
DI = ~ DI
DV = ~ DV
CALL KURENT (SCI, DV)
SCI = SCI - DI
SCTERM = SCI * (1.0 - D/140.0)
SCI = SCI - SCTERM
DI = DI + SCTERM
9 VSHIFT = V + DV
CALL KURENT (VALI, VSHIFT)
I = VALI - DI
RETURN
END

```

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```

RFORTRAN L3,GO,NO,AUP
SUBROUTINE KURENT (I, VSHIFT)
REAL I
COMMON N7A (15), N7A (6), N7B (2), N7C (2), N7C (4),
* N7D (35), N7E, N7F, N7F (9), N7G (3), N7G
COMMON AMPS (200), VOLTS (200), NIV, IC (25), SIC (25), NIC,
* IV (25), DIV (25), NIV, RS
IF (VSHIFT - VOLTS (2)) 21, 21, 24
24 IF (VSHIFT - VOLTS (NIV - 1)) 22, 23, 23
21 NDX = 1
GO TO 25
23 NDX = NIV - 1
GO TO 25
22 LIMLO = 2
LIMHI = NIV - 1
20 LIMD = (LIMHI + LIMLO)/2
IF (VSHIFT - VOLTS (LIMD)) 27, 26, 26
26 LIMLO = LIMD
GO TO 29
27 LIMHI = LIMD
29 IF (LIMHI - LIMLO - 1) 30, 30, 20
28 LIMLO = LIMD
30 NDX = LIMLO
25 I = AMPS (NDX) + (VSHIFT - VOLTS (NDX)) * ((AMPS (NDX + 1) -
* AMPS (NDX))/(VOLTS (NDX + 1) - VOLTS (NDX)))
RETURN
END

```

```

RFORTRAN LS,GO,NS,ADP
SUBROUTINE GETOCV (OI, OV, OCV)
COMMON N7A (13), N7B (6), N7C (2), N7D (2), N7E (2), N7F (2), N7G (2), N7H (2),
* N7I (2), N7J (2), N7K (2), N7L (2), N7M (2), N7N (2), N7O (2), N7P (2), N7Q (2), N7R (2), N7S (2), N7T (2), N7U (2), N7V (2), N7W (2), N7X (2), N7Y (2), N7Z (2),
COMMON AMPS (200), VOLTS (200), NIV, CKOD (50), NLC, DARE (50),
* NIV, RS
NDX = NIV
DO 2 K = 2, NIV
NDX = NIV - K + 1
IF (AMPS (NDX) - OI) 2, 3, 3
2 CONTINUE
OCV = 0.0
RETURN
3 OCV = VOLTS (NDX) - OV - (AMPS (NDX) - OI) * (VOLTS (NDX + 1) -
* VOLTS (NDX)) / (AMPS (NDX + 1) - AMPS (NDX))
RETURN
END

```

```

RFORTRAN LS,GO,ADP
SUBROUTINE KAPPA (RHU,ENN,SN,GAMK,AKAP)
COMMON/BN1/AN1,AK2,H,AKR,R22,X1,X2,FACX1,FACX2,X12,X22,XL,N,NL1,
IF,EN,NL2
DIMENSION F(700)
DATA AN1, H,K,AKR/.0002, 20 .,3332,207./
DATA EN0,SIGR,XLAM/2.32L+32,1.79E-36,3.E-10/
DATA SC,ETA/4.49E-03,.056/
IF(EN0-12345)10,100,10
10 IF(LG1=12345
F(1)=0.
DO 50 I=2,700
Y=I
50 F(I)=ALOG10(Y)+F(I-1)
R2=R*R
R22=2.*R2
HPR2=(H+K)*(H+K)
A2C=3.*ENU*SIGR/8.
XL=XLAM*ALAM/6.2831853
RHOL=1.E+70
ENL=1.E+70
SNL=1.E+70
ULIM=.0174529*45.
100 IF(RHU.EG.RHOL.AND.LIN.EG.ENL) GO TO 150
EN=ENH
X1=RHU
X2=RHU*EN
FACX1=SNL(2./(3.1415927*X1))
FACX2=JUKI(2./(3.1415927*X2))
X12=2.*X1
X22=2.*X2
CALL SIGU(SIGMA)
AK2=SN*SIGMA
CALL INTG(1,0.,ULIM,0.1,0.001)
CALL INTG(2,0.,ULIM,0.12,0.001)
IF(ABS(SN).LE.1.E-07) GO TO 120
CALL INTG(3,0.,ULIM,0.13,0.001)
GO TO 130
120 RI3=0.
130 CALL INTG(4,0.,ULIM,0.14,0.001)
CGAMR=1./R
CALL INTG(5,0.,200.,BI50*CGAMR)
RHOL=RHU
ENL=ENN
GO TO 160
150 IF(SN.EG.SNL) GO TO 200
160 S=SN*SIGMA
A10=EXP(-3C*BI50-3*H)
EX=EXP(-3*H)
A2=A2C*EX*(G1+ETA *BI2)
A3=XL/4.*EX*(BI3/SIGMA+SN*ETA *BI4)
FAC=1.+(1./A10)*(A2+A3)
SNL=SN

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200 CGAM=COS(GAMR)
SGAM=SIN(GAMR)
EL=-R*CGAM+SGR1(HPR2-R2 *SGAM*SGAM)
CGAMR=CGAM/R
CALL INTG(5,0,.EL,B15,CGAMR)
A1=EXP(-JC*B15-S*EL)
AKAP=A1*FAC
RETURN
END

4042-SR

```

RFORTRAN LS,GO,ADP
SUBROUTINE FUNC(I,TYPE,ANS,VAR)
COMMON/UN1/AN1,AN2,H,AUR,R22,AL1,A2,FACAL,FACAL2,X12,X22,XL,N,NL1,
1
IF,EN,NL2
DIMENSION F(700)
IF(I,TYPE,EQ.5) GO TO 5000
COX=CO2(X)
GO TO (1000,2000,3000,4000)I,TYPE
1000 XNUM=(1.-.056/COX)*(1.+COX*COX)*SIN(X)
XLEN=AN1*COX+AN2*(1.-COX)
V=XNUM/XLEN
RETURN
2000 XNUM=(1.+COX*COX)*TAN(X)
XLEN=2.*AN1*COX+AN2*(1.-COX)
V=XNUM/XLEN
RETURN
3000 F1=(1.-.056/COX)
CALL SMALLI(X,ANS1,ANS2)
IF(X)3100,3400,3100
3100 EX=-AN2*(1.-COX)*(206./COX)
F2=(1.-EXP(EX))/(1.-COX)
GO TO 3500
3400 F2=AK2*H
3500 V=F1*F2*(ANS1+ANS2)*SIN(X)
RETURN
4000 CALL SMALLI(X,ANS1,ANS2)
XNUM=(AN1+ANS2)*TAN(X)
XLEN=AK1*COX+AK2*(1.-COX)
V=XNUM/XLEN
RETURN
5000 V=EXP(-AUR*(X*X/R22+X*VAR))
RETURN
END

```

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```

RFUKTRAN LS,000,ADP
  SUBROUTINE INIG(IITYPE,A,0,0,Y,VAR)
    600 CALL FUNC(IITYPE,A,V1,VAR)
    CALL FUNC(IITYPE,0,V2,VAR)
    FAB=V1+V2
    DELX=B-A
    SULD=0.
    SNEW=0.
    YNEW=.9E+50
    N=3
    900 X=A+DELX/2.
    SULD=SULD+SNEW
    SNEW=0.
    YOLD=YNEW
    1000 CALL FUNC(IITYPE,X,V,VAR)
    SNEW=SNEW+V
    X=X+DELX
    IF(X-B)1000,1300,1300
    1300 YNEW=DELA/6.*(FAU+4.*SNEW+2.*SULD)
    1400 IF(ABS(YNEW/YOLD-1.))-1.E-04)1800,18 ,1500
    1500 IF(N-300)1600,1700,1700
    1600 DELX=DELA/2.
    N=2*N-1
    GO TO 900
    1700 WRITE(108,1)IITYPE
    1 FORMATT(11H FLAG FOR 1,12)
    1800 Y=YNEW
    RETURN
    END

```

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```

RFORTRAN LS,GO,ADP
SUBROUTINE SIGMA(SIGMA)
COMMON/UNI/ANI,AN2,H,AOR,R22,X1,X2,FACAL,FACX2,X12,X22,XL,N,NL1,
IF,EINL2
DIMENSION F(700),A(4),RR(4),KKL(4)
DO 100 I=1,4
100 RR(I)=0.
N=0
125 N=N+1
IF (N-350)140,140,130
130 WRITE(6)1,4
4 FORMAT(22H N IS GREATER THAN 350)
STOP
140 DO 200 J=1,4
200 RKL(I)=RKL(I)
CALL VXYZ(V,W,X,Y,Z,A)
N2=2*N+1
220 RKL(I)=RKL(I)+N2*V*V
RKL(2)=RKL(2)+N2*W*W
RKL(3)=RKL(3)+N2*X*Y
RKL(4)=RKL(4)+N2*Z*Z
DO 230 J=1,4
IF(ABS(RR(J)-RKL(J))-1.E-04)230,125,125
230 CONTINUE
SIGMA=XL*(RR(1)+RR(2)+RR(3)+RR(4))
RETURN
END

```

```

RFUNTRAN LS,00,ADP
SUBROUTINE SMALLI(TH,ANS1,ANS2)
COMMON/UNK1/AN1,AK2,H,AGN,R22,X1,X2,FACX1,FACX2,X12,X22,XL,N,NL1,
IF,EN,NL2
DIMENSION F(700),X(4),R(4),RL(4)
CTH=COJ(TH)
DO 175 I=1,4
175 R(I)=0.
ANS1=1.E+50
ANS2=1.E+50
A=1.E+50
R=1.E+50
C=1.E+50
D=1.E+50
IFLA=0
IFLU=0
IFLC=0
IFLD=0
N=0
125 N=N+1
IF(N-350)140,140,130
130 WRITE(6)174)
4 FORMAT(22H N IS GREATER THAN 350)
STOP
140 AL=A
DL=0
CL=C
DL=0
DO 200 I=1,4
200 RL(I)=R(I)
ANS1=ANS1
ANS2=ANS2
CALL VVYZ(V,W,Y,Z,X)
300 IF(N-1)320,310,320
310 ALF=1.
RET=CIH
ALF2=1.
GO TO 350
320 IF(N-2)340,330,340
330 ALF=3.*CIH
ALF1=ALF
RET=6.*CIH*CIH-3.
GO TO 350
340 ALF=((2.*N-1)*ALF1*CIH-N*ALF2)/(N-1)
RET=N*ALF*CIH-(N+1)*ALF1
ALF2=ALF1
ALF1=ALF
350 A=V*DL1+Y*ALF
R=W*RET+Z*ALF
C=V*ALF+Y*DL1
D=W*ALF+Z*DL1
FAC=2*N+1
FAC=FAC/(N*(N+1))

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```

R(1)=RL(1)+FAC*A
R(2)=RL(2)+FAC*B
R(3)=RL(3)+FAC*C
R(4)=RL(4)+FAC*D
ANS1=R(1)*R(1)+R(2)*R(2)
ANS2=R(3)*R(3)+R(4)*R(4)
DO 440 I=1,4
  IF(AUS(RL(1)/R(1)-1.)-.001)440,440,125
440 CONTINUE
480 IF(IFLA-1)495,505,495
495 IF(AUS(A)-AUS(AL))505,505,500
500 IFLA=1
505 IF(IFLB-1)510,525,510
510 IF(AUS(B)-AUS(BL))525,525,520
520 IFLB=1
525 IF(IFLC-1)530,545,530
530 IF(AUS(C)-AUS(CL))545,545,540
540 IFLC=1
545 IF(IFLU-1)550,565,550
550 IF(AUS(U)-AUS(UL))565,565,560
560 IF(LD=1
565 ISUM=IFLA+IFLB+IFLC+IFLU
  IF(ISUM-4)125,600,125
600 ANS1=ANS1L
  ANS2=ANS2L
  RETURN
  END

```

REORTRAN L560U,AUP

SUBROUTINE VMYZIV(W,Y,Z,X)
COMMON/DN1/AN1,AN2,H,AUR,RZ2,X1,X2,FACX1,FACX2,X12,X22,XL,N,NL1,

IF,EN,NL2

DIMENSION F(700),X(4)

NL1=(N-1)/2+1.

NL2=N/2+1.

CALL DGGP(1,X)

DX1=X(1)

DX2=X(2)

QX1=X(3)

QX2=X(4)

CALL DGGP(2,X)

GX1=X(1)

GX2=X(2)

PX1=X(3)

PX2=X(4)

ANG=X1 +N*1.5707963

SANG=SIN(ANG)

CANG=COS(ANG)

ELX1=(1-FACX1)*(PX1*(SANG+CANG/X12)+GX1*(CANG-SANG/X12)-GX1*CANG

1 +DX1*SANG)

AKX1=FACX1*(PX1*CANG-GX1*SANG)

CALL HJ(HX1,AJX1,X1,N,PA1,GA1,GA1,DX1,FACX1)

CALL HJ(HX2,AJX2,X2,N,PA2,GA2,GA2,DX2,FACX2)

FKX1=JGR1(1.5707963*X1)

FKX2=JGR1(1.5707963*X2)

XU=(-1)**(N+1)*FNA1

FX1=X0*AKX1

GX1=X0*(ELX1+AKX1/X12)

PS1X1=FKX1*AJX1

PS1X2=FKX2*AJX2

EX1=FKX1*(HX1+AJX1/X12)

EX2=FKX2*(HX2+AJX2/X22)

EM=EN*PS1X1*EX2-EX1*PS1X2

S=FN*EX2*FX1-PS1X2*GX1

T=EN*EX1*PS1X2-PS1X1*EX2

U=EN*PS1X2*GX1-EX2*FX1

DEN=EM*EM+S*S

VE=(-EM)*EM/DEN

W=FM*S/DEN

DEN=1*1+U*U

Y=(-1)*1/DEN

Z=1*U/DEN

RETURN

END

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RFORTPAN L3,GO,ADP
SUBROUTINE HJTHA,AJA,ASN,P,U,U,FAC)

X2=2.*X

ANG=X-N*1.5707963

SANG=Sin(ANG)

CANG=Cos(ANG)

HX=FAC*(P*(CANG-SANG/X2)-Q*(SANG+CANG/X2)+G*SANG+D*CANG)

AJX=FAC*(P*SANG+Q*CANG)

RETURN

END

4042-SR

RFORTRAN LS,GO,ADP
SUBROUTINE DDDP(I,YP,X)
COMMON/CK1/AK1,AK2,H,AOR,R22,X1,X2,FACX1,FACX2,X12,X22,XL,N,NL1,

IF,EN,NL2
DIMENSION X(4),F(700)

DO 50 I=1,4

50 X(I)=0.

GO TO (60,70)I,YP

60 NL=NL1

GO TO 100

70 NL=NL2

100 DO 1000 I=1,NL

90 GO TO (110,120)I,YP

110 I1=2*I-1

GO TO 120

120 I1=2*(I-1)

120 IF((I/2)*2-1)140,130,140

130 NML=1

GO TO 150

140 NML=-1

150 NML2=N+1

XNML2=F(NML2)

IDEN1=I1-1

GO TO(300,500)I,YP

300 IF(IDEN1)370,360,370

360 XDEN1=0.

GO TO 380

370 XDEN1=F(IDEN1)

380 IDEN2=N-1

IF(IDEN2+1)400,723,400

400 IF(IDEN2)420,410,420

410 XDEN2=0.

GO TO 700

420 XDEN2=F(IDEN2)

GO TO 700

500 IDEN2=N-1

IF(IDEN2)520,510,520

510 XDEN2=0.

GO TO 530

520 XDEN2=F(IDEN2)

530 IF(IDEN1+1)550,723,550

550 XDEN1=F(IDEN1)

700 XDEN1=XNML2-XDEN1-XDEN2

725 DLI1=11*ALOG10(2.*X1)

DEN2=11*ALOG10(2.*X2)

X(1)=X(1)+NML*EXP(2.*D258509*(XND-DLI1-ALOG10(X1)))

X(2)=X(2)+NML*EXP(2.*D258509*(XND-DEN2-ALOG10(X2)))

728 IF(1)740,730,740

730 X(3)=1.

X(4)=1.

GO TO 1000

740 YI=1

XDEN3=XDEN1+ALOG10(YI)

800 XND=XNUM2-XDEN2
X(3)=X(3)-NUM1*EXP(2.30258509*(XND-DEN1))
X(4)=X(4)-NUM1*EXP(2.30258509*(XND-DEN2))

1000 CONTINUE
RETURN
END

7254EL

REFORTRAN L3, GUNINS, ADP
SUBROUTINE KAUAT(I, I2, L, EM)

INTEGER I2

IF (L-1) 10, 20, 10

20 EM=1.0

RETURN

10 EM= 1.0-0.21*(1-I2)/365.

IF (EM) 30, 40, 40

30 EM=0.0

40 RETURN

END

7254EL

4042-SR

RFORTRAN LS,GO,NS,ADP
SUBROUTINE CEDAT(X,Y,LZ,LL)
FL=1
RETURN
END

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7254EL

```

RFORTRAN LS,GO,NS,ADP
SUBROUTINE D3TAT (AGUIN, TAU, KFLAG, AT)
COMMON /BERNIE/ RHO, ENN, EN
COMMON /DUSTY/ A, ETA, GRU
COSX = SIN (AGUIN)
IF (KFLAG) 1, 2, 1
1 AT = 1.0
RETURN
2 H = 0.96E12 * (1.0 + 0.343E-8/A) * EN * TAU * A ** 3
AT = EXP (-ETA * H/COSX) * (1.0 - GRU * (1.0 - EXP (-H/A)))
RETURN
END

```

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7254JLNU

RFORTRAN L3,GO,N5,ADP
SUBROUTINE JTIME(MO,DAY,IYR,JDAY)
INTEGER DYOFM ,S,MU,DAY,IYR,JDAY

DIMENSION DYOFM (11)

DYOFM(1)=31

DYOFM(2)=28

DYOFM(3)=31

DYOFM(4)=30

DYOFM(5)=31

DYOFM(6)=30

DYOFM(7)=31

DYOFM(8)=31

DYOFM(9)=30

DYOFM(10)=31

DYOFM(11)=30

S=(IYR -1967)*365

**ADD A DAY FOR EACH INTERVING LEAP YEAR

IF(1968-IYR) 100,130,110

100 DO 1201=1968,IYR+4

IF(1-IYR)120,130,150

120 S=S+1

110 IM=1964

115 IF(IM-IYR)150,140,160

160 S=S-1

IM=IM-4

**ADD ONE TO DAYS IF YEAR IS A LEAP YEAR AND MU IS 3

GO TO 115

130 IF(MO-3)150,160,180

180 S=S+1

GO TO 150

140 IF(MO-3)155,150,150

155 S=S-1

**ADD NUMBER OF DAYS IN YEAR UP TO SPECIFIED MONTH(MU)

150 I=MO-1

IF(1)170,190,170

170 DO 175 J=1,I

175 S=S+DYOFM(J)

**ADD NUMBER OF DAYS + SUBTRACT 110 DAYS (1-1 TO 4-20)

190 JDAY = S+DAY-110

RETURN

END

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```

RFORTRAN LS,00,NS,ADP
SUBROUTINE TEMP (LAT,TIME,FGS,PHI,IMPGU,PIEMP)
C LAT=LATITUDE
C TIME=TIME FROM DAWN
C FGS=FGS IN MILLIWATTS/CM SQ
C PHI = PHI NAUTICAL
C TIME=TIME FROM SET = 0. IN INITIALIZATION AREA OF MAIN PROGRAM
C PIEMP=PIEMP TEMPERATURE IN DEGREES C
REAL LAT

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4042-SR

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DIMENSION TSO(13),TAO(13),T40(13),T40(13)
IF (IMPGU)100,10,100

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10 IMPG=1.
C TEMP. AT 90 DEG. LATITUDE - DEG. F
T90=-109.3

```

```

C SURFACE TEMPS AT 0 DEG. LAT. - DEG. F
TSO(1)=-171.5
TSO(2)=-123.

```

```

TSO(3)=-66.
TSO(4)=-42.
TSO(5)= 4.
TSO(6)= 23.

```

```

TSO(7)= 58.
TSO(8)= 46.
TSO(9)= 18.
TSO(10)= 15.

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```

TSO(11)= 46.
TSO(12)= 73.
TSO(13)= 92.5

```

```

C ATMOSPHERIC TEMPS AT 0 DEG. LAT. DEG. F
TAO(1)=-162.5
TAO(2)=-158.5

```

```

TAO(3)=-150.
TAO(4)=-135.
TAO(5)=-107.5
TAO(6)=- 82.

```

```

TAO(7)=- 76.5
TAO(8)=- 70.
TAO(9)=- 73.
TAO(10)=- 80.5

```

```

TAO(11)=- 90.
TAO(12)=- 98.5
TAO(13)=-107.

```

```

C SURFACE TEMPS AT 40 DEG. LAT. DEG. F
T40(1)=-198.5
T40(2)=-172.

```

```

T40(3)=-113.
T40(4)=- 57.
T40(5)=- 42.
T40(6)=- 2.

```

```

T40(7)= 14.5
T40(8)= 26.
T40(9)= 40.
T40(10)= 56.

```

TS40(11)=-81.
TS40(12)=-103.
TS40(13)=-119.
C ATMOSPHERIC TEMPS AT 40 DEG. LAT.

TA40(1)=-192.
TA40(2)=-189.
TA40(3)=-182.
TA40(4)=-170.
TA40(5)=-149.
TA40(6)=-119.5
TA40(7)=-94.5
TA40(8)=-64.
TA40(9)=-90.
TA40(10)=-104.
TA40(11)=-116.
TA40(12)=-126.
TA40(13)=-134.5

C CALC. SURF. AND ATMOS. TEMPS AT 0 AND 40 DEG. LAT. FOR GIVEN TIME

100 ITW=TIME
PART=TIME-116.

111=116+1
112=116+2

IF(112-13)500,500,510

500 TS40(1)=TS40(11)+PART*(TS40(112)-TS40(111))
TS40(1)=TS40(11)+PART*(TS40(112)-TS40(111))

GO TO 520

510 TS01F=(TIME-12.1)/14.

TS40(1)=(TS40(11)-TS40(13))*TS01F + TS4 (13)
TS40(1)=(TS40(11)-TS40(13))*TS01F + TS4 (13)

C CALC. SURF. AND ATMOS. TEMPS AT GIVEN LAT.

520 ALAT=ABS(LAT)

IF(ALAT-40.1150,150,150,200

150 IF(112-13)500,500,540

530 TS01F=TS0(11)+PART*(TS0(112)-TS0(111))

TS01F=TS0(11)+PART*(TS0(112)-TS0(111))

GO TO 550

TS01F=TS0(11)-TS40(1)

ARG=ALAT/40.8125/100

DTA=TS01F*COS(ARG)

DT3=TS01F*COS(ARG)

TS=TS40(1)+DT3

TA=TA40(1)+DTA

GO TO 225

540 TS01F=(TS0(11)-TS0(13))*TS01F+TS0(13)

TS01F=(TS0(11)-TS0(13))*TS01F+TS0(13)

GO TO 550

200 TS=TS40(1)+(ALAT-40.)*(190-TS40(1))/50.

TA=TA40(1)+(ALAT-40.)*(190-TA40(1))/50.

C CONVERT TO DEGS. N

225 TA=(TA-32.)/1.8+273.1

TS=(TS-32.)/1.8+273.1

C CONVERT TO METER IN SEC.

FGS=FGS*.0064516

```

C      CALC. PANEL TEMP. USING NEWTON - RAPHSOON METHOD
C      C=F03*(.729*CO3(PH1)+.09)+3.111E-12*TA**4+.31.11E-12*IS**4+
1      .00454*IS
      TNEW=IA+10.
250  F=66.22E-12*INLW**4+.00454*INLW-C
      FAB=ABS(F)
      IF (FAB<.1) 400,400,300
300  INLW=INLW-F/(2.0480E-10*INLW**3+.00454)
      GO TO 250
C      CONVERT TO DEG. C
400  PTEMP=INLW-273.1
      RETURN
      END

```

4042-SR

KLCD 160*,)UNJAI,1F4L10**,)JL,7*

RRUN

RDATA

0.22

0.0

0.085

2.48

2.214

2.19

0.0

100.0

55.0

100.0

0

.1119999

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72

.1119681 .132500
.1119679 .135000
.1119678 .138000
.1119676 .141000
.1119674 .144000
.1119672 .147000
.1119670 .150000
.1119668 .153000
.1119666 .156000
.1119663 .159000
.1119660 .162000
.1119657 .165000
.1119654 .168000
.1119651 .171000
.1119647 .174000
.1119643 .177000
.1119639 .180000
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.1119629 .186000
.1119624 .189000
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.1119612 .195000
.1119606 .198000
.1119600 .201000
.1119591 .204000
.1119583 .207000
.1119574 .210000
.1119569 .213000
.1119555 .216000
.1119544 .219000
.1119533 .222000
.1119520 .225000
.1119507 .228000
.1119492 .231000
.1119477 .234000
.1119460 .237000
.1119443 .240000
.1119423 .243000
.1119403 .246000
.1119381 .249000
.1119367 .252000
.1119351 .255000
.1119334 .258000
.1119314 .261000
.1119294 .264000
.1119279 .267000
.1119262 .270000
.1119243 .273000
.1119220 .276000
.1119195 .279000
.1119166 .282000
.1119144 .285000
.1119118 .288000

.1119127 .2910000
.1119062 .2940000
.1118992 .2970000
.1118917 .3000000
.1118837 .3030000
.1118750 .3060000
.1118657 .3090000
.1118557 .3120000
.1118450 .3150000
.1118334 .3180000
.1118210 .3210000
.1118077 .3240000
.1117934 .3270000
.1117780 .3300000
.1117614 .3330000
.1117437 .3360000
.1117246 .3390000
.1117041 .3420000
.1116820 .3450000
.1116584 .3480000
.1116329 .3510000
.1116055 .3540000
.1115762 .3570000
.1115447 .3600000
.1115108 .3630000
.1114743 .3660000
.1114352 .3690000
.1113721 .3720000
.1113480 .3750000
.1112954 .3780000
.1112472 .3810000
.1111912 .3840000
.1111510 .3870000
.1110962 .3900000
.1109967 .3930000
.1109220 .3960000
.1108417 .3990000
.1107555 .4020000
.1106928 .4050000
.1106033 .4080000
.1104563 .4110000
.1103413 .4140000
.1102178 .4170000
.1100851 .4200000
.1099425 .4230000
.1097593 .4260000
.1096247 .4290000
.1094478 .4320000
.1092578 .4350000
.1090536 .4380000
.1088342 .4410000
.1085989 .4440000
.1083452 .4470000

•1083731	•4200300
•1077607	•4200300
•1074665	•4200300
•1071282	•4200300
•1067662	•4200300
•1064763	•4200300
•1059577	•4200300
•1055678	•4200300
•1050244	•4200300
•1045050	•4200300
•1039469	•4200300
•1033472	•4200300
•1027029	•4200300
•1020100	•4200300
•1012668	•4200300
•1004876	•4200300
•0996089	•4200300
•0988862	•4200300
•0976943	•4200300
•0960290	•4200300
•0954851	•4200300
•0942554	•4200300
•0929341	•4200300
•0915144	•4200300
•0909890	•4200300
•0882501	•4200300
•0865890	•4200300
•0846767	•4200300
•0826039	•4200300
•0807754	•4200300
•0781323	•4200300
•0756105	•4200300
•0729003	•4200300
•0699893	•4200300
•0668613	•4200300
•0633002	•4200300
•0598080	•4200300
•0560005	•4200300
•0518292	•4200300
•0473395	•4200300
•0425463	•4200300
•0373746	•4200300
•0318179	•4200300
•0286474	•4200300
•0194223	•4200300
•0125395	•4200300
•0051335	•4200300
-0.0028240	•4200300
-0.0113740	•4200300
-0.0205007	•4200300

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75

1 ELECTRO-OPTICAL SYSTEMS JPL COUN 952035
 2 02/01/74 03/05/74 02/01/74 10002 40.00.0 0.0 20. 100. 0 0.0 5 50
 3 6.20 1.5 0.5E-09 1.0E 00 0.5

1 ELECTRO-OPTICAL SYSTEMS JPL CONTR 952035
 2 02/01/74 03/05/74 02/01/74 10/02 40.00.0 0.0 20. 100. 0 0.0 5 50
 3 6.28 1.5 3.0E 15 0.5E-09 1.0E 06 0.5

REOD
 REOD
 RFIN

4.4 TYPICAL PRINTOUT OF SAMPLE PROBLEM

The data cards listed in 4.3 were used to run a sample problem. The associated printout follows here. The first page reiterates the input configuration data. The second page is representative of the output to be expected, but does not include the total printout from this run; only the first two pages are shown.

MARS SOLAR POWER ANALYSIS PROGRAM

JPL CONTR 952035

RUN TITLE... ELECTRO-OPTICAL SYSTEMS

ANALYSIS PERIOD FROM 2/ 1/74 TO 3/ 5/74

SPACECRAFT LANDING DATE= 2/ 1/74

ATTENUATION FACTORS USED-

SMALL A = .50E-09 KM

RHB = 6.28

LARGE N = 1.50

SMALL N = .30E 16 KM(-3)

ZETA = .10E 07 KM(-1)

NU = .500

RADIATION MODEL FLAG = 2

DUST STORM IN PROGRESS

SPACECRAFT POSITION-

LATITUDE 40.0 DEG

PANEL INCLINED FROM

HORIZONTAL BY

DIRECTION OF INCLINE

(AZIMUTH)

SOLAR PANEL AREA 20.0 SQ.FT.

REQUIRED POWER OUTPUT 100.0WATTS

DEFINED HORIZON 0.0 DEG

PRINTER CONTROL-

PRINT, SUMMARY EVERY 5 DAY(S)

PRINT DETAIL EVERY .493 HRS DURING DAYLIGHT

(ONLY ON SUMMARY DAYS)

SOLAR PANEL OUTPUT BASED ON MARTNER 1969 S/N 2, 2X2 CM SOLAR CONVERTER I-V CURVE, N/P .018 IN, 2 8HM.CM/
MISCELLANEOUS POWER LOSSES 5.01 PER CENT

RUN TITLE... ELECTRO-OPTICAL SYSTEMS JPL CNTR 952035

LATITUDE = 40.0 DEG

TIME UNIT MAX REQUIRED ACCUM SI AT TRANSMISSION S.I. AT PANEL CELL RADIAT SUN SUN INCIDENCE

OF DAY CELL PANEL AREA ENERGY UPPER FACTORS TEMP DEGRAD ELEV AZIM- ANGLE

(HOUR) POWER POWER (WATT- ATMS (WATT(SQ.FT.) HOURS) ATM0 CLOUD DUST (MW/SQ. CM) (CEN) RADIAT BEFFR FACTOR UTH BN

(MW) (WATT(SQ.FT.) HOURS) (MW/SQ. CM) (CEN) RADIAT (DG) (DEG) PANEL

MARS DAY = 1, EARTH TIME = .0000 DAYS PAST HOUR 0.0 OF 2 / 174 GMT

*** DAWN = SUN 6N PANEL

6.28	.00	.0045970.4	.0	57.72	.00	1.00	1.00	.03	.00	.114.4	.00	1.00	90.0	92.4	90.0
7.00	.03	.1414151.4	.0	57.71	.01	1.00	1.00	.51	.07	.113.2	.03	1.00	81.9	99.3	81.9
7.50	.16	.74 2684.6	.3	57.71	.02	1.00	1.00	1.44	.34	.111.9	.17	1.00	76.4	104.3	76.4
8.00	.46	2.11 947.9	1.0	57.71	.05	1.00	1.00	2.85	.93	.109.9	.48	1.00	71.1	109.4	71.1
8.50	.87	4.03 496.3	2.5	57.71	.08	1.00	1.00	4.55	1.86	.79.5	.92	1.00	65.9	114.9	65.9
9.00	1.39	6.42 311.4	5.2	57.71	.11	1.00	1.00	6.31	3.07	.64.0	1.47	1.00	60.9	121.0	60.9
9.50	1.95	8.97 223.0	9.0	57.70	.14	1.00	1.00	7.96	4.42	.51.3	2.05	1.00	56.3	127.6	56.3
10.00	2.49	11.49 174.1	14.2	57.70	.16	1.00	1.00	9.41	5.78	.42.4	2.62	1.00	52.1	134.9	52.1
10.50	2.97	13.71 145.8	20.5	57.70	.18	1.00	1.00	10.60	7.04	.35.1	3.13	1.00	48.4	143.1	48.4
11.00	3.37	15.53 128.8	27.9	57.70	.20	1.00	1.00	11.52	8.08	.29.8	3.55	1.00	45.4	152.2	45.4
11.50	3.63	16.74 119.5	36.0	57.70	.21	1.00	1.00	12.14	8.85	.25.3	3.82	1.00	43.3	162.2	43.3
12.00	3.75	17.29 115.7	44.6	57.69	.22	1.00	1.00	12.48	9.27	.21.7	3.95	1.00	42.0	172.9	42.0
12.50	3.72	17.16 116.6	53.3	57.69	.22	1.00	1.00	12.53	9.33	.18.7	3.92	1.00	41.9	183.8	41.9
13.00	3.55	16.35 122.3	61.7	57.69	.21	1.00	1.00	12.28	9.02	.16.4	3.74	1.00	42.8	194.5	42.8
13.50	3.34	15.41 129.8	69.7	57.69	.20	1.00	1.00	11.75	8.36	.23.0	3.52	1.00	44.6	204.8	44.6
14.00	3.11	14.33 139.5	77.2	57.69	.19	1.00	1.00	10.92	7.39	.33.2	3.27	1.00	47.4	214.2	47.4
14.50	2.70	12.47 160.4	84.0	57.68	.17	1.00	1.00	9.81	6.20	.44.0	2.85	1.00	50.9	222.7	50.9
15.00	2.27	10.47 190.9	89.8	57.68	.15	1.00	1.00	8.44	4.85	.63.4	2.39	1.00	54.9	230.3	54.9
15.50	1.62	7.47 267.8	94.3	57.68	.12	1.00	1.00	6.85	3.48	.67.1	1.71	1.00	59.4	237.1	59.4
16.00	1.01	4.68 427.7	97.3	57.68	.09	1.00	1.00	5.11	2.22	.70.4	1.07	1.00	64.3	243.3	64.3
16.50	.53	2.42 825.0	99.1	57.68	.06	1.00	1.00	3.38	1.19	.73.5	.55	1.00	69.4	249.0	69.4
17.00	.20	.94 2131.6	100.0	57.68	.03	1.00	1.00	1.84	.49	.76.3	.21	1.00	74.7	254.2	74.7
17.50	.05	.22 9000.1	100.3	57.67	.01	1.00	1.00	.74	.13	.78.9	.05	1.00	80.1	259.2	80.1
18.00	.00	.0204189.5	100.3	57.67	.00	1.00	1.00	.18	.01	.81.2	.00	1.00	85.6	264.0	85.6

*** DUSK = SUN CUT OFF OF PANEL ***

18.00	.00	.0022503.6	100.3	57.67	.00	1.00	1.00	.03	.00	.82.8	.00	1.00	90.0	267.7	90.0
-------	-----	------------	-------	-------	-----	------	------	-----	-----	-------	-----	------	------	-------	------

*** MAXIMUM POWER POINT ***

12.12	3.76	17.32	115.5	46.6	57.69	.22	1.00	1.00	12.52	9.32	.20.9	3.96	1.00	41.9	175.3
-------	------	-------	-------	------	-------	-----	------	------	-------	------	-------	------	------	------	-------

SUMMARY = 6.28 HRS, HOURS OF SUN = 12.12, MAX OUTPUT = 17.32W, ENERGY FOR DAY = 100.3WH, ENERGY TO DATE = 100.3WH

NEAR DAWN (18.9 DEG.)

PERIOD OF MAX. POWER

NEAR DUSK (9.9 DEG.)

SHORT CIRCUIT CURRENT

MAXIMUM POWER VOLTAGE

OPEN CIRCUIT VOLTAGE

8.8

8.63

8.63

8.63

8.63

8.63

8.63

8.63

SECTION 5

CONCLUSIONS AND RECOMMENDATIONS

The overall Planetary Solar Array Development program undertaken by EOS during the past several years has defined the important aspects of the Mars hard and soft lander solar power problem. The work reported here has provided a means for computing the probable panel output for a large number of configurations of the cells and panels, for various geometries and, most important, under the expected environmental conditions.

Earlier reports have described designs which would successfully meet many difficult constraints of power, weight, and structural integrity. In addition, important test data were generated, showing solar cell shock response as a function of radius of curvature and of loading up to 5000 g's. The entire body of work contains many important theoretical and engineering advances in the areas of orbital dynamics, light scattering, the physics of blown sand and dust, solar cell models, and computer programming, as well as a rather complete structural analysis of the statics and dynamics of hard and soft landing and panel deployment.

It is recommended that the computer program described in this report be modified, either for the JPL computer facility or for the EOS machine, to produce plots of the important parameters computed during the Mars day. These graphs would then be used to define envelopes of worst- and best-case conditions which could be published in the form of a design handbook for Mars lander vehicles. In the event it should be desirable to produce these plots manually, the computer program in its present form will be adequate, and only the input parameters remain to be specified.

As a further recommendation, it appears feasible to adapt the present computer program to calculate the power output of an orbiting panel.

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7254-Q-3 (3 Volumes)	15 Apr 68
7254-Final	28 Feb 69