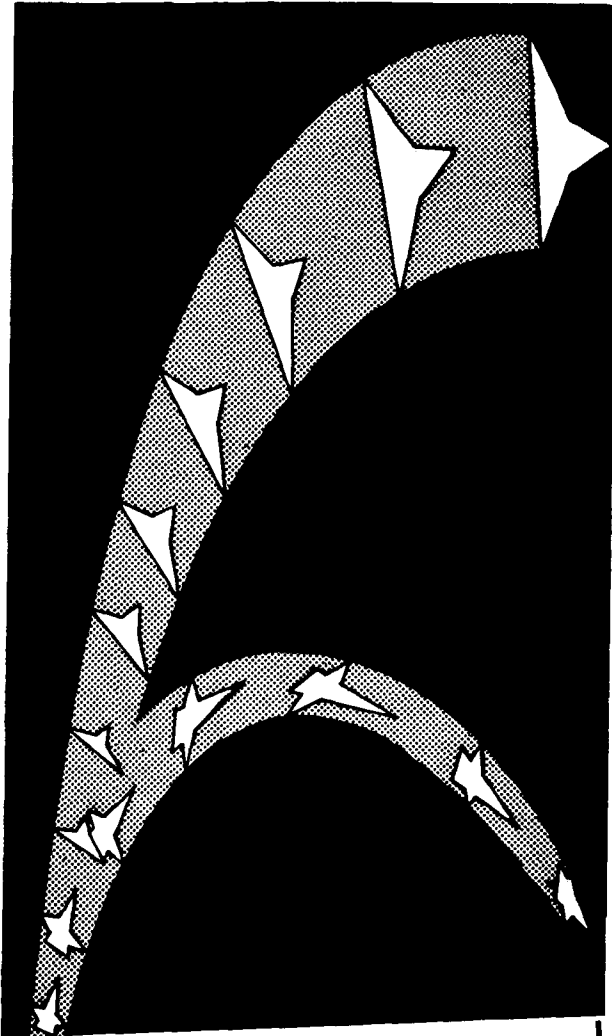


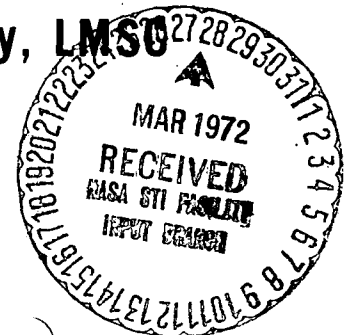


—SPACE SHUTTLE—

**EXPERIMENTAL INVESTIGATIONS
FOR BASE DRAG REDUCTION
ON A 0.015 SCALE MODEL
MSFC PROPOSED SPACE SHUTTLE
BOOSTER AT MACH NUMBERS
FROM 0.40 TO 1.10**

by

Dale Bradley, LMSO



N72-19895

**FOOT
TUNNEL**

Unclas
21086

CSCL 22B G3/31

**Lockheed
Missile and Space Corp.**

**SADSAC SPACE SHUTTLE
AEROTHERMODYNAMIC
DATA MANAGEMENT SYSTEM**

**CONTRACT NAS8-4016
MARSHALL SPACE FLIGHT CENTER**

SPACE DIVISION  **CHRYSLER
CORPORATION**

This document should be
referenced as
NASA CR-120,030

SADSAC/SPACE SHUTTLE
WIND TUNNEL TEST DATA REPORT

CONFIGURATION: MSFC Parametric Booster (.015 Scale Model)

TEST PURPOSE: Base Drag Reduction Investigation

TEST FACILITY: Cornell Aero. Lab 8 x 8 Foot Transonic Wind Tunnel

TESTING AGENCY: LMSC - Huntsville

TEST NO. & DATE: CAL 18-063, 1-2 Nov., 1971 (OCC. HRS. 28)

FACILITY COORDINATOR: Robert J. Dennis

PROJECT ENGINEER(S): Dale Bradley, LMSC

DATA MANAGEMENT SERVICES

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RELEASE APPROVAL: for N. D. Kemp, Supervisor
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CONTRACT NAS 8-4016

AMENDMENT 153

DRL 184-58

This report has been prepared by Chrysler Corporation Space Division under a Data Management Contract to the NASA. Chrysler assumes no responsibility for the data presented herein other than its display characteristics.

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EXPERIMENTAL INVESTIGATIONS FOR BASE DRAG REDUCTION ON A
.015 SCALE MODEL MSFC PROPOSED SPACE SHUTTLE BOOSTER
AT MACH NUMBERS FROM 0.40 TO 1.10

By Dale Bradley

SUMMARY

A 0.015-scale model of a modified version of the MDAC Space Shuttle Booster was tested in the Cornell Aeronautical Laboratory 8 x 8 Foot Transonic Wind Tunnel during November 1971 to obtain force, static stability, and control effectiveness data. The objective of this test was the reduction of cruise ($M = 0.4$) base drag by the use of base flaps, base vents, elevon deflection and base flow from a plenum mounted forward of the base heat shield. Transonic data were also obtained to determine the aerodynamic characteristics of the new base shape. Six component aerodynamic force and moment data were recorded over an angle of attack range from -4° to 20° at 0° sideslip and over a sideslip range from -6° to 6° at 0° , 6° and 15° angle of attack. Mach number varied from 0.4 to 1.10 at a constant R of 2.0×10^6 per unit length.

NOMENCLATURE

(General)

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
α	ALPHA	angle of attack, angle between the projection of the wind X_w -axis on the body X, Z-plane and the body X-axis; degrees
β	BETA	sideslip angle, angle between the wind X_w -axis and the projection of this axis on the body X-Z-plane; degrees
ψ	PSI	yaw angle, angle of rotation about the body Z-axis, positive when the positive X-axis is rotated toward the positive Y-axis; degrees
ϕ	PHI	roll angle, angle of rotation about the body X-axis, positive when the positive Y-axis is rotated toward the positive Z-axis; degrees
ρ		air density; K_g/m^3 , slugs/ft ³
a		speed of sound; m/sec, ft/sec
V		speed of vehicle relative to surrounding atmosphere; m/sec, ft/sec
q	$Q(\text{PSI})$ $Q(\text{PSF})$	dynamic pressure; $1/2\rho V^2$, psi, psf
M	MACH	Mach number; V/a
RN/L	RN/L	Reynolds number per unit length; million/ft
p		static pressure; psi
P		total pressure; psi
C_p	CP	pressure coefficient; $(p-p_\infty)/q$

NOMENCLATURE (Continued)

Reference & C. G. Definitions

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
S		wing area; m^2 , ft^2
S	SREF	reference area; m^2 , ft^2
$\bar{c}$		wing mean aerodynamic chord or reference chord; m, ft, in (see l_{ref} or LREF)
l_{ref}	LREF	reference length; m, ft, in.; (see $\bar{c}$)
b_{ref}	BREF	wing span or reference span; m, ft, in
A_b		base area; m^2 , ft^2 , in^2
c. g.		center of gravity
MRP	MRP	abbreviation for moment reference point
	XMRP	abbreviation for moment reference point on X-axis
	YMRP	abbreviation for moment reference point on Y-axis
	ZMRP	abbreviation for moment reference point on Z-axis

NOMENCLATURE (Continued)

Axis System General

SYMBOL

DEFINITION

F

force; F, lbs

M

moment; M, in-lb

Subscript

Definition

N

normal force

A

axial force

L

lift force

D

drag force

Y

force or moment about the Y axis

Z

moment about the Z axis

X

moment about the X axis

s

stability axis system

w

wind axis system

ref

reference conditions

∞

free stream conditions

t

total conditions

b

base

NOMENCLATURE (Continued)
Body & Stability Axis System

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
<u>Body Axis System</u>		
C_N	CN	normal force coefficient; F_N/qS
C_A	CA	axial force coefficient; F_A/qS
C_{A_b}	CAB	base axial force coefficient; $\begin{bmatrix} -1 \\ \end{bmatrix} \begin{bmatrix} (p_b - p_\infty)/q \end{bmatrix} (A_b/S)$
C_{A_f}	CAF	forebody axial force coefficient; $C_A - C_{A_b}$
C_n	CYN	yawing moment coefficient; $M_Z/qS b_{ref}$
C_l	CBL	rolling moment coefficient; $M_X/qS b_{ref}$
<u>Common to Both Axis Systems</u>		
C_m	CLM	pitching moment coefficient; $M_Y/qS l_{ref}$
C_y	CY	side force coefficient; F_Y/qS
<u>Stability Axis System</u>		
C_L	CL	lift force coefficient; F_L/qS
C_D	CD	drag force coefficient; F_D/qS
C_{D_b}	CDB	base drag coefficient
C_{D_f}	CDF	forebody drag coefficient; $C_D - C_{D_b}$
C_n	CLN	yawing moment coefficient; $M_{Z,s}/qS b_{ref}$
C_l	CSL	rolling moment coefficient; $M_{X,s}/qS b_{ref}$
L/D	L/D	lift-to-drag ratio; C_L/C_D
L/D_f	L/DF	lift to forebody drag ratio; C_L/C_{D_f}

NOMENCLATURE (Continued)

Surface Definitions

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
1_t	HORIZT	horizontal tail incidence; positive when trailing edge down; degrees
δ		symmetrical surface deflection angle; degrees; positive deflections are:
	AILRON	aileron - total aileron deflection; (left aileron - right aileron)/2
	CANARD	canard - trailing edge down
	ELEVON	elevon - trailing edge down
	ELEVTR	elevator - trailing edge down
	FLAP	flap - trailing edge down
	RUDDER	rudder - trailing edge to the left
	SPOILER	spoiler - trailing edge down
	TAB	tab - trailing edge down with respect to control surface
δ		antisymmetrical surface deflection angle, degrees; positive trailing edge down:
	AIL-L	left aileron - trailing edge down
	AIL-R	right aileron - trailing edge down
	ELVN-L	left elevon - trailing edge down
	ELVN-R	right elevon - trailing edge down
	SPLR-L	left spoiler - trailing edge down
	SPLR-R	right spoiler - trailing edge down

<u>SURFACE SUBSCRIPTS</u>	<u>DEFINITION</u>
a	aileron
b	base
c	canard
e	elevator or elevon
f	flap
r	rudder or ruddervator
s	spoiler
t	tail

CONFIGURATIONS INVESTIGATED

The 0.015 scale model MSFC Space Shuttle Booster is a modified version of the MDAC (256-14) booster configuration and consists of the following components:

B ₄	Modified MDAC 256-15 fuselage
B ₅	Same as B ₄ except has body flaps
B ₆	Same as B ₄ except has slots in body
B ₇	Same as B ₄ except has rocket nozzles removed
W ₃	Wings
V ₁	Wing tip mounted vertical fin
C ₂	Small canard
F ₂	Trailing edge flap on canard

The configurations tested are shown in Table I. Table III gives pertinent dimensional information on each of the above components.

Grit was used on the model to insure boundary layer transition. Figure 12 depicts the manner in which it was applied and the grit size used as determined by the method of Braslow and Knox.

TEST FACILITY DESCRIPTION

The 8-Foot Transonic Wind Tunnel was placed in operation in December of 1956 as the result of modernizing the 12-Foot Variable Density Wind Tunnel to extend its operation through the transonic range. The tunnel has a perforated throat and an auxiliary pumping system for plenum pumping. The continuous circuit tunnel is capable of operating from $1/6$ to $2-1/2$ atmospheres total pressure, thereby providing a wide range of test Reynolds numbers as well as Mach numbers. The range of operating pressures is necessarily limited by the total power available at the higher Mach numbers. Pumping the tunnel to these conditions is done by four centrifugal compressors for above one atmosphere testing and by seven compressors for below one atmosphere. Evacuation of the tunnel to 800 psf total pressure can be accomplished by use of the auxiliary compressor from atmospheric pressure. This procedure takes approximately 8 minutes. Consequently, at least an initial expenditure of time is necessary to bring the tunnel to the desired operating conditions. During model changes, two gate valves isolate the test section from the tunnel proper making it necessary to bring only the test sphere to atmospheric conditions. By careful planning of the test program it is then possible to reduce pumping time to a minimum.

The test section of the tunnel is a removable cart. In many instances this feature permits the installation of a model prior to testing resulting in a saving of tunnel time. Three carts are in active use: a sting cart for the testing of sting-mounted, full-span models, a reflection plane cart for use with semi-span reflection plane models and the fairing cart for full-span models mounted from a plate.

Low speed airflow calibrations have been performed for free-stream velocities from 5 to 90 feet per second. Velocities in this range are steady and can be set accurately using a fixed main drive blade angle and varying the rpm. Low speed tests may be run within the operating tunnel densities of $1/6$ of an atmosphere to 2.5 atmospheres.

More explicit details of the tunnel and its operational characteristics can be found in the 8-Foot Transonic Wind Tunnel Report WTO-300 at Cornell Aeronautical Laboratory.

DATA REDUCTION

The six component force and moment data recorded by the balance were corrected for weight tares, tunnel flow angularity, sting and balance deflection and reduced to coefficient form using the following reference values.

$$S_{REF} = \text{wing planform area} = 1.355 \text{ ft}^2$$

$$l_{REF} = \text{unmodified body length} = 3.453 \text{ ft}$$

$$b_{REF} = l_{REF}$$

Moments are referenced to a center of gravity location 2.594 ft aft of the nose and 0.01875 ft above fuselage centerline (see Figure 2).

Corrections to axial force measurement for the effects of base pressure were made utilizing the following equations:

$$C_{AB} = (C_{P_1} + C_{P_2} + C_{P_3} + C_{P_4}/4) A_1/S_{REF} + (C_{P_5} + C_{P_6} + C_{P_7} + C_{P_8}/4) A_2/S_{REF}$$

where

$C_{P_1}, C_{P_2}, C_{P_3}, C_{P_4}$ are pressure coefficients of pressures measured on the heat shield

$C_{P_5}, C_{P_6}, C_{P_7}, C_{P_8}$ are pressure coefficients of pressures measured at the nozzle exit

$$A_1 = \text{heat shield area} = 0.1629 \text{ ft}^2$$

$$A_2 = \text{nozzle exit area} = 0.0792 \text{ ft}^2$$

TABULATED DATA LISTING

A tabulated data listing, consisting of all aero data sets, both original and those created in arriving at the plotted material to be presented subsequently, is available as an addendum to this report. The tabular listing is made up in two sections:

- (a) a brief summary list of all data sets containing the identifier, the descriptor, and the resident dependent variables.
- (b) a full list of all data sets containing all resident or selected aerodynamic coefficients of the data sets as well as the above mentioned information.

The listing is currently sent on limited distribution to the following organizations:

NASA AMES	Mr. V. Stevens
CAL	Mr. R. J. Dennis
LMSC	Mr. C. Donald Andrews

If copies of this listing are desired, please contact the above or the cognizant SADSAC personnel who, for this data, is:

Mr. Albert D. Martin
Department 2780
Chrysler Corporation Space Division
New Orleans, La. 70129

(504) 255-2304

TEST CAL 8x8 DATA SET COLLATION SHEET

TABLE I

☐ PRETEST
☒ POSTTEST

DATA SET IDENTIFIER	CONFIGURATION	SCHD.		PARAMETERS/VALUES				NO. OF RUNS	MACH NUMBERS (OR ALTERNATE INDEPENDENT VARIABLE)							
		A	B	1	2	3	4		0.1	0.2	0.3	0.4	0.5	0.6	0.7	0.8
R09001	B4C2F2W3V1	A	0					1	7	74	78	80	83	86		
0000		0	A					1	3	75	79	81	85	87		
0000		6							73	76	80	82	84	88		
0001		15						1	12	77	81	83	85	89		
0000	B5C2F2W3V1	A	0					1	14							
0000		0	A					1	16							
0000		6						1	17							
0000		15						1	15							
0000	B6C2F2W3V1	A	0					1	20							
0100		0	A					1	21							
0110		6						1	22							
0120		15						1	23							
0130	B4C2F2W3V1	A	0	10	0			1	24							
0140		0	A					1	25							
0150		6						1	26							
0160		15						1	27							
0170		A	0	-10				1	41							
0180		0	A					1	15							
0190		6						1	16							
0200		15						1	41							

COEFFICIENTS: 1 7 13 19 25 31 37 43 49 55 61 67 75 76

a or b SCHEDULES

4A - 0 1 4 2 6 8 10 15 20
8A - 0 1 3 4 6 8 10 15 20

12

POSTTEST

7576

```

=>IDPVAR(1)|IDPVAR(2)|NDV

```

NASA-MSFC-HAF

TEST CHL 8x6' DATA SET COLLATION SHEET

13

☐ PRETEST
☐ POSTTEST

DATA SET IDENTIFIER	CONFIGURATION	SCHD.		PARAMETERS/VALUES			NO. of RUNS	MACH NUMBERS (OR ALTERNATE INDEPENDENT VARIABLE)									
		A	B	5	5	10		0.40									
R09011	84C2F2 W3V1	A	0	-10	20	-	1	36									
012		0	A				1	37									
013		6					1	38									
014		15					1	39									
015		A	0	+10	10		1	40									
016		0	A				1	41									
017		6					1	42									
018		15					1	43									
019		A	0	-10			1	44									
050		0	A				1	45									
051		6					1	46									
052		15					1	47									
053		A	0	-20	0		1	48									
054		0	A				1	49									
055		6					1	50									
056		15					1	51									
057		A	0	0	0	353	1	52									
058		0	A				1	53									
059		6					1	54									
060		15					1	55									

1 7 13 19 25 31 37 43 49 55 61 67 7576

COEFFICIENTS:
a or b
SCHEDULES

NASA-MSEC-MAF

TEST CONDITIONS
TEST CAL 8 x 8' - 063

BALANCE UTILIZED: CAL-TASK-MK XIX

**COEFFICIENT
TOLERANCE:**

$$\begin{array}{r} +.0230 \\ +.0115 \\ +.0019 \\ +.0014 \\ +.0006 \\ +.0004 \end{array}$$

Coefficient tolerance is for $M = 1.0$, $q = 385$ psf.

TABLE III. MODEL COMPONENT DESCRIPTION

MODEL COMPONENT: BODY - B₄ (Base Flow)

GENERAL DESCRIPTION: Modified MDAC 256-14 Body ($\lambda = 3.638'$) with Large
Base Area to Shield Engines. There is a Non-Metric Base Flow Plenum Installed.

DRAWING NUMBER:

CAL W14-0047B

DIMENSIONS:

FULL SCALE

MODEL SCALE

Length

242 ft3.638 ft

Max. Width

37.7 ft0.566 ft

Max. Depth

34.0 ft0.510 ft

Fineness Ratio

Area

Max. Cross-Sectional

1076 ft²0.242 ft²

Planform

Wetted

Base

1076 ft²0.242 ft²

TABLE III. MODEL COMPONENT DESCRIPTION (Continued)

MODEL COMPONENT: W₃

GENERAL DESCRIPTION: This wing is basically the same as W₁ except it was modified near the root chord. This modification results in a slight decrease in wing exposed area due to the flare of the body at the base (B₄)

DRAWING NUMBER:

NSRDC A152024-1DIMENSIONS:FULL-SCALEMODEL SCALETOTAL DATA

Area

Planform

6020 ft²1.355 ft²

Wetted

--

--

Span (equivalent)

120.67 ft

1.81 ft

Aspect Ratio

2.42

2.42

Rate of Taper

--

--

Taper Ratio

.547

.547

Dihedral Angle, degrees

7°40', 15°

7°40', 15°

Incidence Angle, degrees

0°, 3°

0°, 3°

Aerodynamic Twist, degrees

--

--

Toe-In Angle

0°

0°

Cant Angle

--

--

Sweep Back Angles, degrees

--

--

Leading Edge

40°

40°

Trailing Edge

19°31'

19°31'

0.25 Element Line

--

--

Chords:

Root (Wing Sta. 0.0)

64.55 ft

.969 ft

Tip, (equivalent)

35.4 ft

.53 ft

MAC

--

--

Fus. Sta. of .25 MAC

--

--

W.P. of .25 MAC

--

--

B.L. of .25 MAC

--

--

Airfoil Section

Root

64A010

64A010

Tip

64A010

64A010

EXPOSED DATA

Area

3960 ft².891 ft²

Span, (equivalent)

86.8 ft

1.30 ft

Aspect Ratio

1.89

1.89

Taper Ratio

.627

.627

Chords

Root

56.4 ft

.845 ft

Tip

35.4

.53 ft

MAC

--

--

Fus. Sta. of .25 MAC

--

--

W.P. of .25 MAC

--

--

B.L. of .25 MAC

--

--

TABLE III. MODEL COMPONENT DESCRIPTION (Continued)

MODEL COMPONENT: BODY - B5 (Base Flaps)

GENERAL DESCRIPTION: Same as B4 except flaps on the top and bottom (60" x 318" full scale) and in the sides (60" x 212" full scale). Flaps are deflected in toward the booster main engine nozzles.

DRAWING NUMBER: CAL W14-0046B

DIMENSIONS:	FULL SCALE	MODEL SCALE
Length	<u>242 ft</u>	<u>3.638 ft</u>
Max. Width	<u>37.7 ft</u>	<u>0.566 ft</u>
Max. Depth	<u>34.0 ft</u>	<u>0.510 ft</u>
Fineness Ratio	<u>--</u>	<u>--</u>
Area		
Max. Cross-Sectional	<u>1076 ft²</u>	<u>0.242 ft²</u>
Planform	<u>--</u>	<u>--</u>
Wetted	<u>--</u>	<u>--</u>
Base	<u>1076 ft²</u>	<u>-.242 ft²</u>

TABLE III. MODEL COMPONENT DESCRIPTION (Continued)

MODEL COMPONENT: BODY - B6 (Base Venting)

GENERAL DESCRIPTION: Same as B4 except slots have been cut in top and bottom
(60" x 318" full scale) and in the sides (60" x 212" full scale) just aft of
the heat shield (STA. 3861 Rel. to Nose at STA. 1025)

DRAWING NUMBER: CAL W14-0045B

DIMENSIONS:	FULL SCALE	MODEL SCALE
Length	<u>242 ft</u>	<u>3.638 ft</u>
Max. Width	<u>37.7 ft</u>	<u>0.566 ft</u>
Max. Depth	<u>34.0 ft</u>	<u>0.510 ft</u>
Fineness Ratio	<u>--</u>	<u>--</u>
Area		
Max. Cross-Sectional	<u>1076 ft²</u>	<u>0.242 ft²</u>
Planform	<u>--</u>	<u>--</u>
Wetted	<u>--</u>	<u>--</u>
Base	<u>1076 ft²</u>	<u>0.242 ft²</u>

TABLE III. MODEL COMPONENT DESCRIPTION (Continued)

MODEL COMPONENT: BODY - B7

GENERAL DESCRIPTION: Same as B4 except the booster main rocket nozzles
(except four partial nozzles around the sting) were removed.

DRAWING NUMBER: _____

DIMENSIONS:	FULL SCALE	MODEL SCALE
Length	<u>242 ft</u>	<u>3.638 ft</u>
Max. Width	<u>37.7 ft</u>	<u>0.566 ft</u>
Max. Depth	<u>34.0 ft</u>	<u>0.510 ft</u>
Fineness Ratio	<u>--</u>	<u>--</u>
Area	<u> </u>	<u> </u>
Max. Cross-Sectional	<u>1076 ft²</u>	<u>0.242 ft²</u>
Planform	<u>--</u>	<u>--</u>
Wetted	<u>--</u>	<u>--</u>
Base	<u>1076 ft²</u>	<u>0.242 ft²</u>

TABLE III. MODEL COMPONENT DESCRIPTION (Continued)

MODEL COMPONENT: V1GENERAL DESCRIPTION: Booster Wing Tip Vertical Fin

(Dimensions given are for each tip)

DRAWING NUMBER: NRDC A152027-1.4DIMENSIONS:FULL-SCALEMODEL SCALETOTAL DATA

Area		
Planform	1010 ft ²	.226 ft ²
Wetted	3 --	--
Span (equivalent)	38.80 ft	.582 ft
Aspect Ratio	1.5	1.5
Rate of Taper	--	--
Taper Ratio	.467	.467
Diehedral Angle, degrees	65°	65°
Incidence Angle, degrees	3°	3°
Aerodynamic Twist, degrees	0	0
Toe-In Angle	0	0
Cant Angle	0	0
Sweep Back Angles, degrees		
Leading Edge	40°	40°
Trailing Edge	19°31'	19°31'
0.25 Element Line	--	--
Chords:		
Root (Wing Sta. 0.0)	35.35 ft	.530 ft
Tip, (equivalent)	16.50 ft	.247 ft
MAC	--	--
Fus. Sta. of .25 MAC	--	--
W.P. of .25 MAC	--	--
B.L. of .25 MAC	--	--
Airfoil Section		
Root	64A010	64A010
Tip	64A009	64A009

EXPOSED DATA

Area	--	--
Span, (equivalent)	--	--
Aspect Ratio	--	--
Taper Ratio	--	--
Chords		
Root	--	--
Tip	--	--
MAC	--	--
Fus. Sta. of .25 MAC	--	--
W.P. of .25 MAC	--	--
B.L. of .25 MAC	--	--

TABLE III. MODEL COMPONENT DESCRIPTION (Continued)

MODEL COMPONENT: C2GENERAL DESCRIPTION: Booster Small Canard

DRAWING NUMBER:

NSRDC A152018-5DIMENSIONS:FULL-SCALEMODEL SCALETOTAL DATA

Area

Planform

2235 ft²0.503 ft²

Wetted

--

--

Span (equivalent)

72.6 ft

1.09 ft

Aspect Ratio

2.36

2.36

Rate of Taper

--

--

Taper Ratio

.118

.118

Dihedral Angle, degrees

0

0

Incidence Angle, degrees

3°, 15°

3°, 15°

Aerodynamic Twist, degrees

0

0

Toe-In Angle

--

--

Cant Angle

--

--

Sweep Back Angles, degrees

53°

53°

Leading Edge

0

0

Trailing Edge

--

--

0.25 Element Line

--

--

Chords:

Root (Wing Sta. 0.0)

55 ft

.825 ft

Tip, (equivalent)

6.47 ft

.097 ft

MAC

--

--

Fus. Sta. of .25 MAC

--

--

W.P. of .25 MAC

--

--

B.L. of .25 MAC

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Airfoil Section

Root

64A010

64A010

Tip

64A010

64A010

EXPOSED DATA

Area

750 ft²0.169 ft²

Span, (equivalent)

38.8 ft

0.581 ft

Aspect Ratio

2

2

Taper Ratio

0.2

0.2

Chords

Root

32.2 ft

0.484 ft

Tip

6.46 ft

0.097 ft

MAC

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Fus. Sta. of .25 MAC

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W.P. of .25 MAC

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B.L. of .25 MAC

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TABLE III. MODEL COMPONENT DESCRIPTION (Concluded)

MODEL COMPONENT: F2 with Canard C2

GENERAL DESCRIPTION: Booster Canard Trailing Edge Flap (40% Chord -
Exposed Area)

DRAWING NUMBER: NSRDC A152018-5

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area	<u>300 ft²</u>	<u>0.0676 ft²</u>
Span (equivalent)	<u>38.8 ft</u>	<u>0.581 ft</u>
Inb'd equivalent chord	<u>12.9 ft</u>	<u>0.1936 ft</u>
Outb'd equivalent chord	<u>2.59 ft</u>	<u>0.0388 ft</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u>0.4</u>	<u>0.4</u>
At Outb'd equiv. chord	<u>0.4</u>	<u>0.4</u>
Sweep Back Angles, degrees		
Leading Edge	<u>28°</u>	<u>28°</u>
Tailing Edge	<u>0</u>	<u>0</u>
Hingeline	<u>--</u>	<u>--</u>
Area Moment (Normal to hinge line)	<u>--</u>	<u>--</u>

TABLE IV. INDEX OF MODEL FIGURES

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10	Body Flaps (Top View)	37
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TABLE V. INDEX TO DATA FIGURES

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Comparison of New Base Shape With Old Base Shape-Longitudinal Characteristics	A, C	Configuration	1-20
Comparison of New Base Shape With Old Base Shape-Lateral-Directional Characteristics	B, D	Configuration	21-22
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Effect of Booster Base Shape on Longitudinal Characteristics	A	Configuration	27-29
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Elevon Control Effectiveness - Longitudinal Characteristics	A	Elevator Deflections	31-33 35-37 39-41
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Canard Control Effectiveness - Lateral- Directional Characteristics	B	Canard Deflections	46
Effect of Varying Engine Weight Flow Rate on Longitudinal Characteristics	E	Configuration	47-48

TABLE V. INDEX TO DATA FIGURES
(Continued)

TITLE	PLOTTED		PAGES
	COEFFICIENTS SCHEDULE	CONDITIONS VARYING	
Aerodynamic Characteristics as a Function of Elevator Deflection	F	ALPHA	49-52
Aerodynamic Characteristics as a Function of Canard Deflection	F	ALPHA	53-56

PLOTTED COEFFICIENTS SCHEDULE:

- A) CL vs. ALPHA, C_{LM}; CA, CAB, CN vs. ALPHA
L/D, CD vs. ALPHA; CL vs. CD
- B) C_Y, C_{YN}, C_{BL} vs. BETA
- C) DC_{LM}DA, C_{LA}LEA vs. MACH
C_{DA}FO, C_{AB}AF_O, C_{AA}FO vs. MACH
- D) C_YBETA, DC_{YN}DB, DC_{BL}DB vs. MACH
- E) C_{LM}, CL, CD, L/D, CN, CA, CAB vs. BSFLOW
- F) C_{LM}, CL, CD, CAB vs. ANGLE

MODEL FIGURES

- Notes:
1. Positive directions of force coefficients moment coefficients, and angles are indicated by arrows.
 2. For clarity, origins of wind and stability axes have been displaced from the center of gravity.

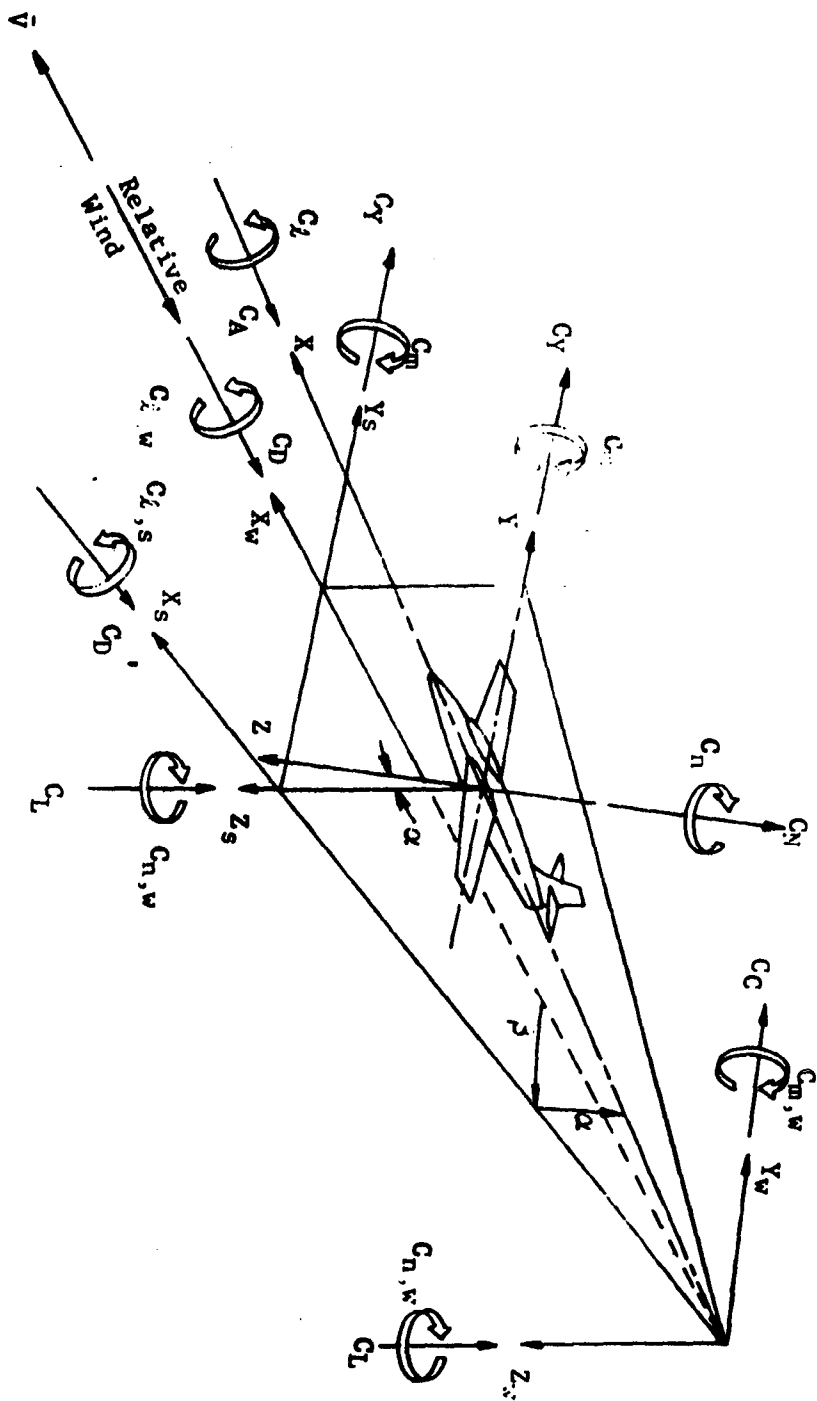


Figure 1. Axis systems, showing direction and sense of force and moment coefficients, angle of attack, and sideslip angle

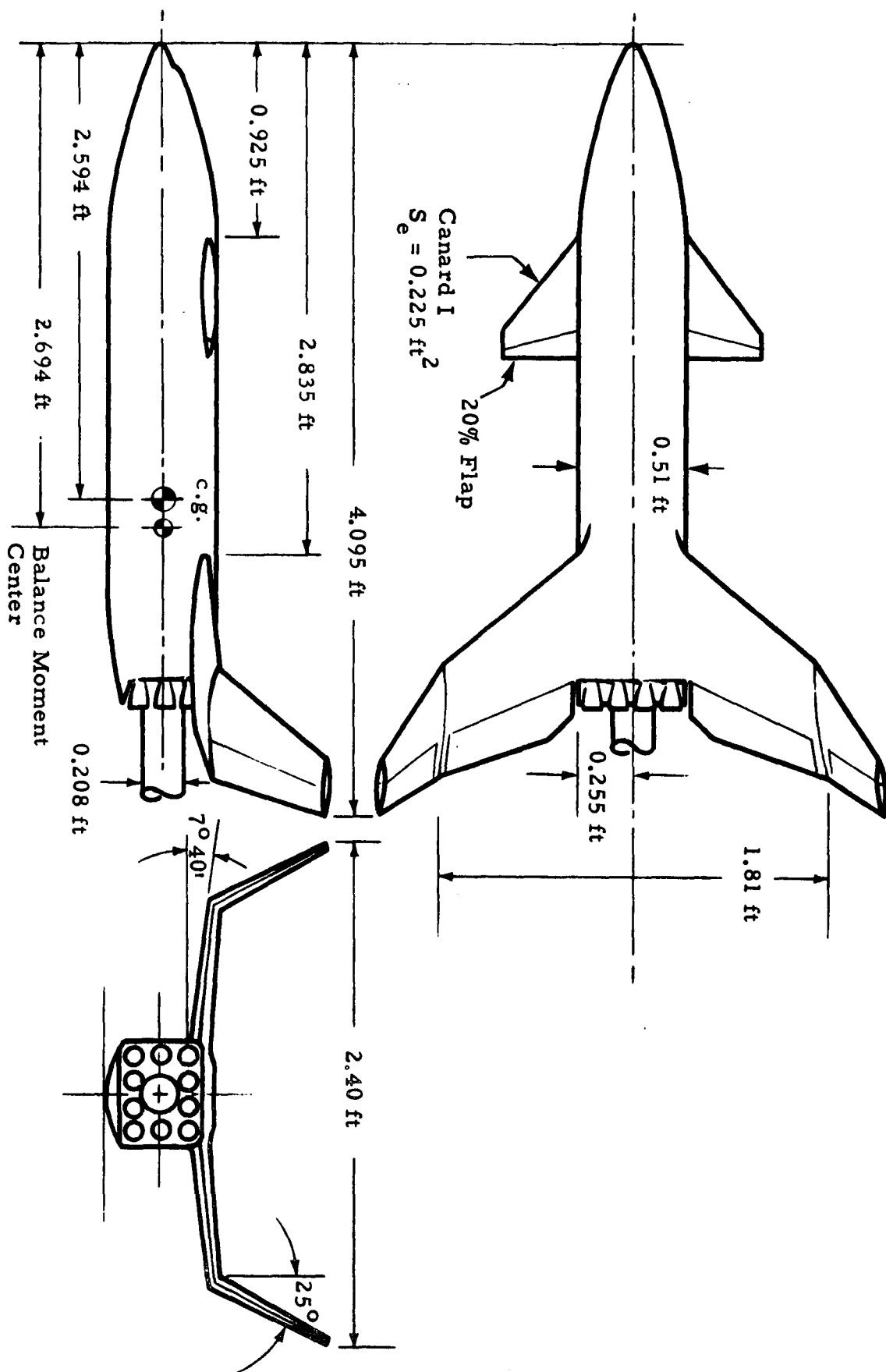


Fig. 2 - General Arrangement of the Baseline Model

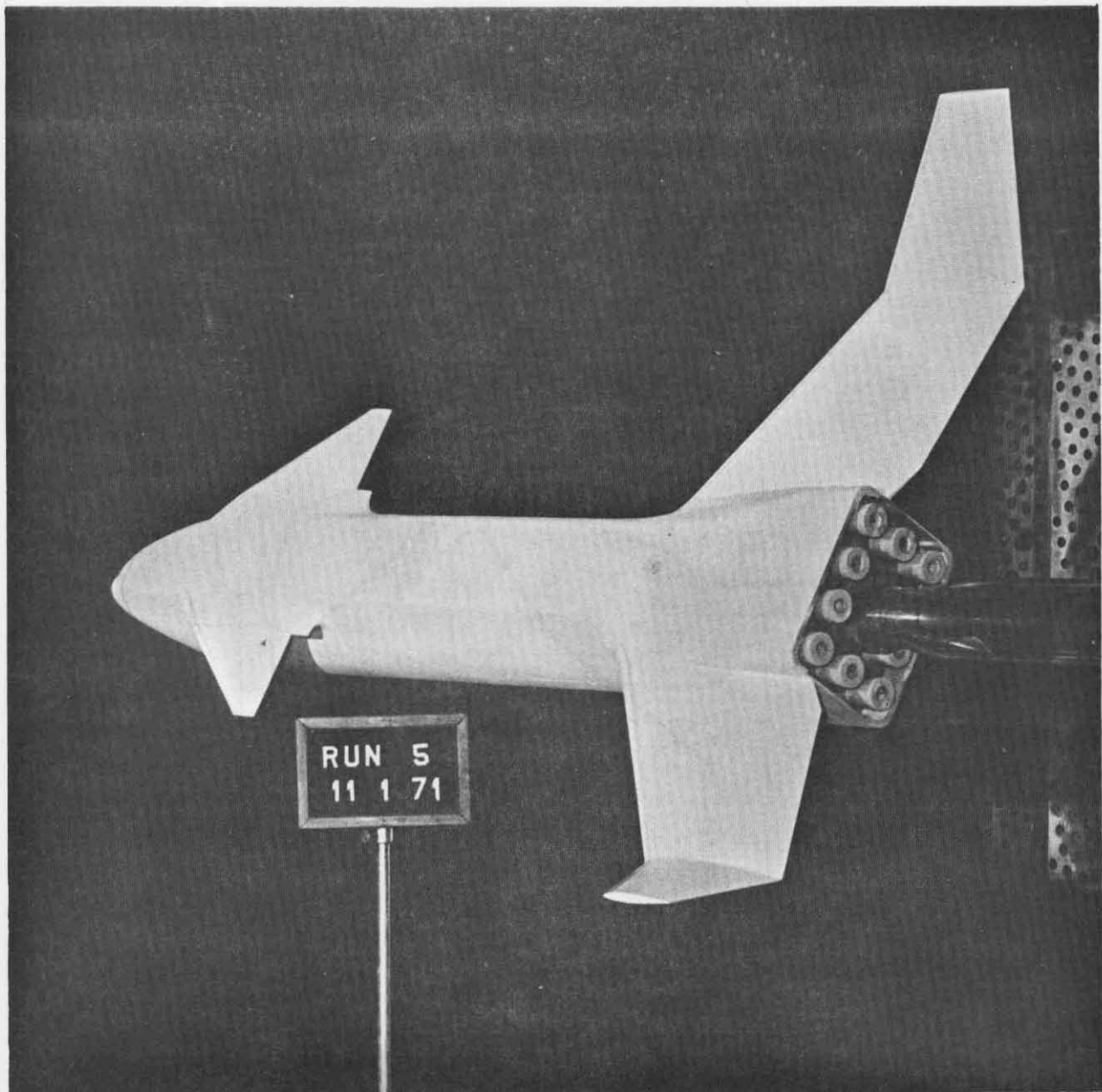


FIGURE 3. PHOTOGRAPH OF $B_4C_2F_2W_3V_1$ INSTALLED IN CAL 8 x 8 FOOT TUNNEL

THIS PHOTOGRAPH CONTAINS PROPRIETARY INFORMATION.
IT MAY NOT BE REPRODUCED OR DISSEMINATED WITHOUT
PRIOR PERMISSION OF THE TEST SPONSOR.

THIS PHOTOGRAPH ORIGINATED AT
CORNELL AERO. LAB., BUFFALO, N.Y.

NOVEMBER 1971

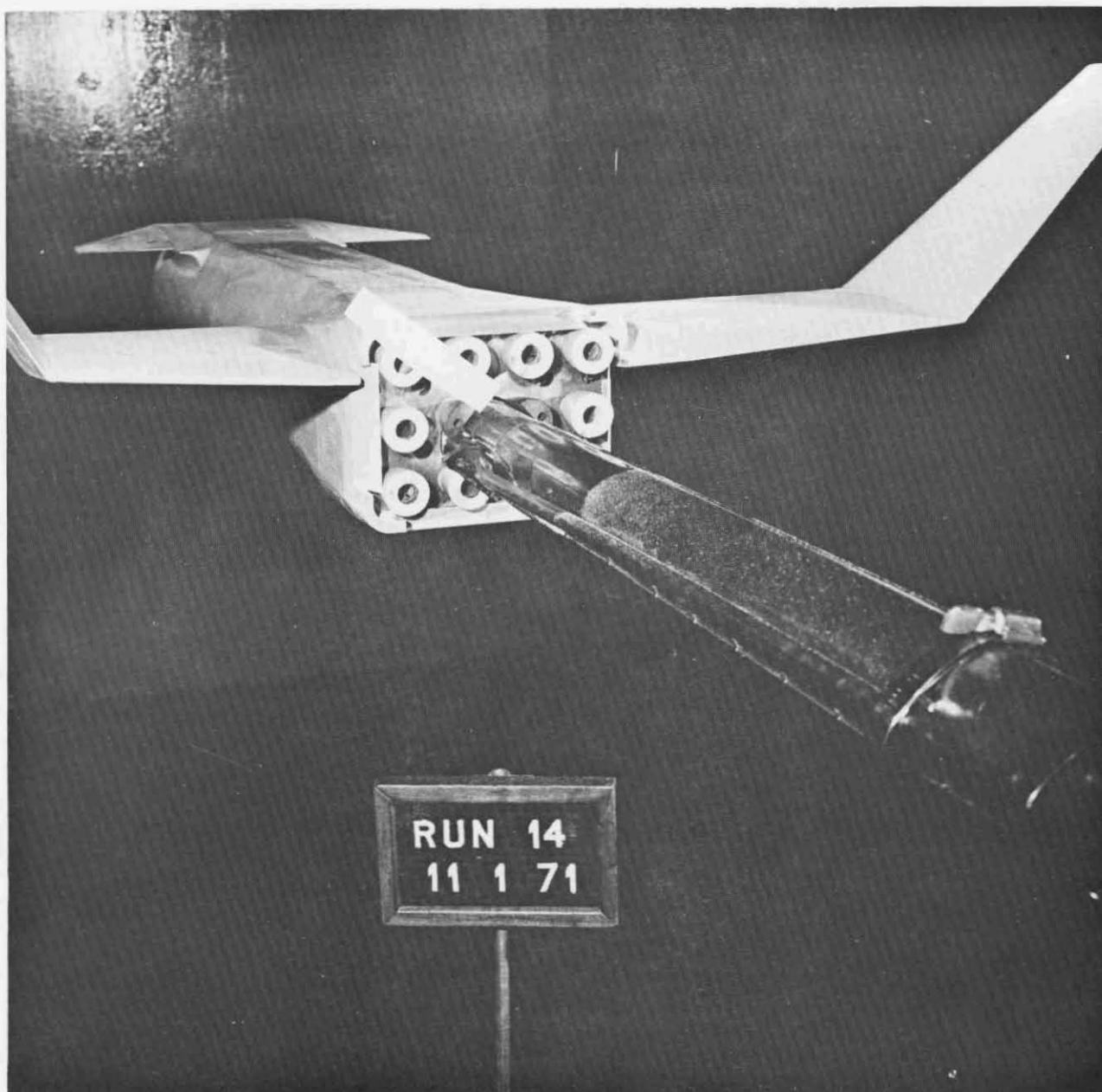


FIGURE 4. PHOTOGRAPH OF B₅C₂F₂W₃V₁ INSTALLED IN CAL 8 x 8 FOOT TUNNEL

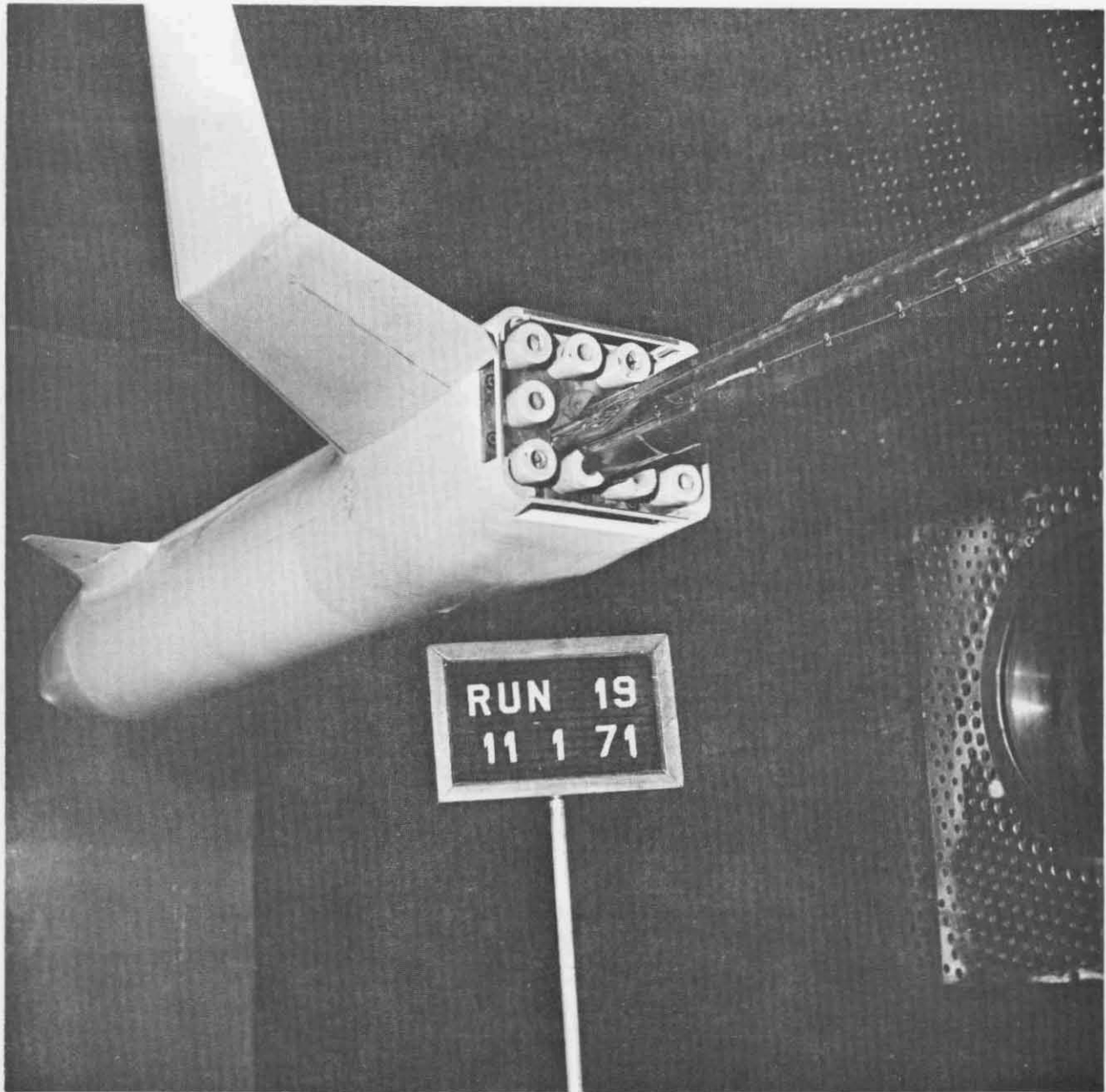


FIGURE 5. PHOTOGRAPH OF $B_6C_2F_2W_3V_1$ INSTALLED IN CAL 8 x 8 FOOT TUNNEL

THIS PHOTOGRAPH CONTAINS PROPRIETARY INFORMATION.
IT MAY NOT BE REPRODUCED OR DISSEMINATED WITHOUT
PRIOR PERMISSION OF THE TEST SPONSOR.

- Note: (1) All stations are relative to 1025 which is the station at the nose.
- (2) All dimensions are given in full scale; model scale is 0.015.

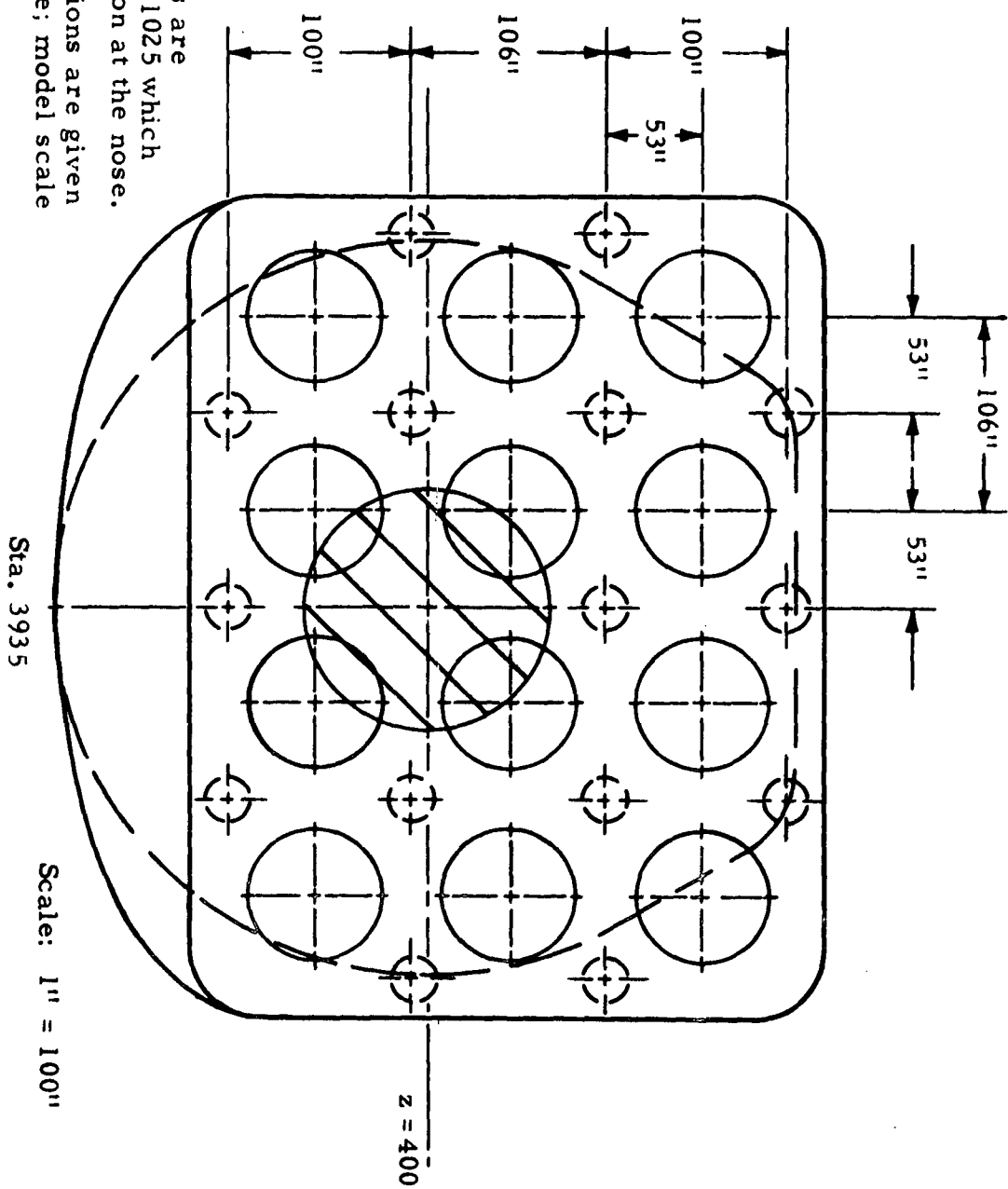


Fig. 6 - Base Plenum Orifice Location (End View)

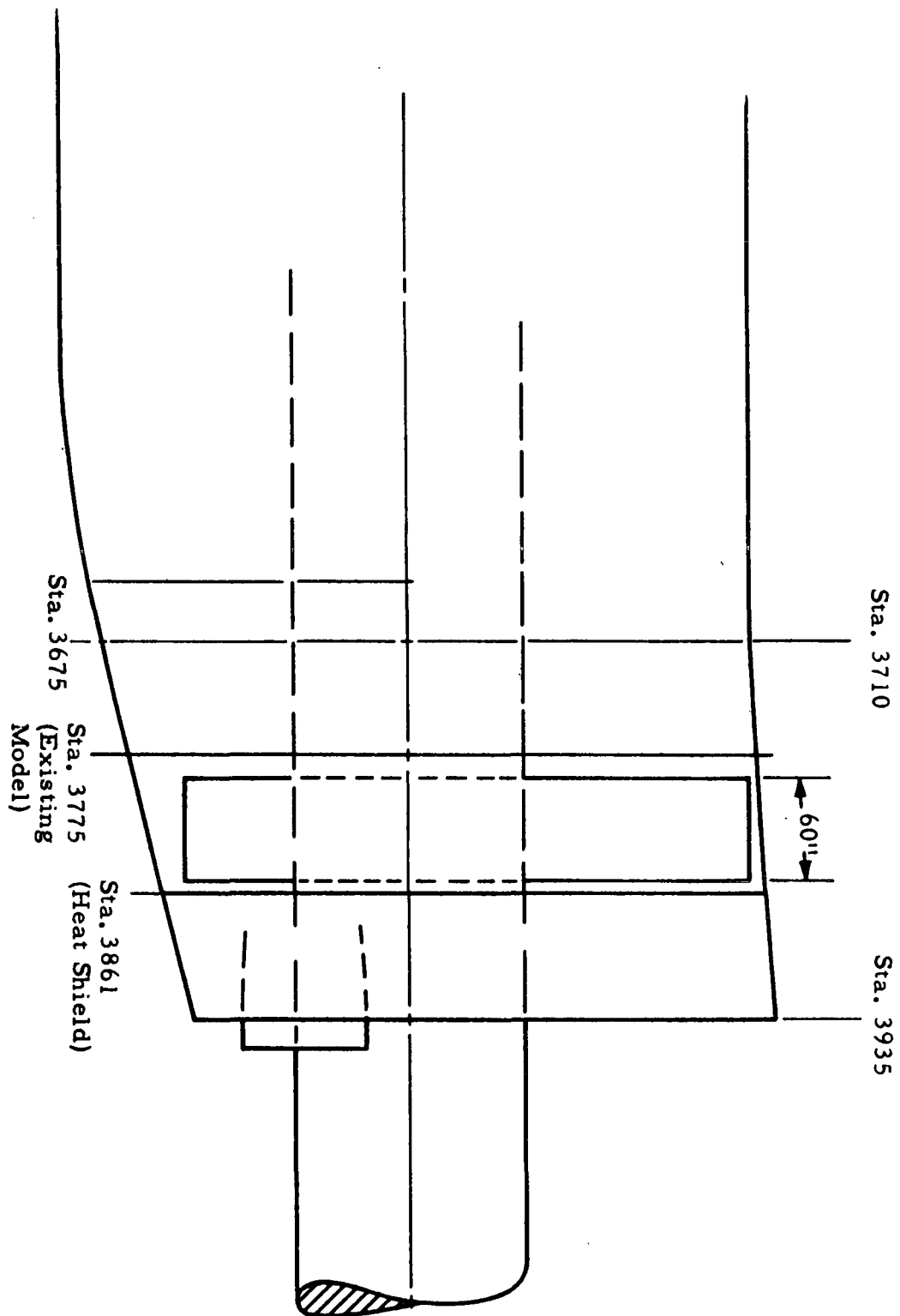


Fig. 7 - Base Plenum Orifice Location (Side View)

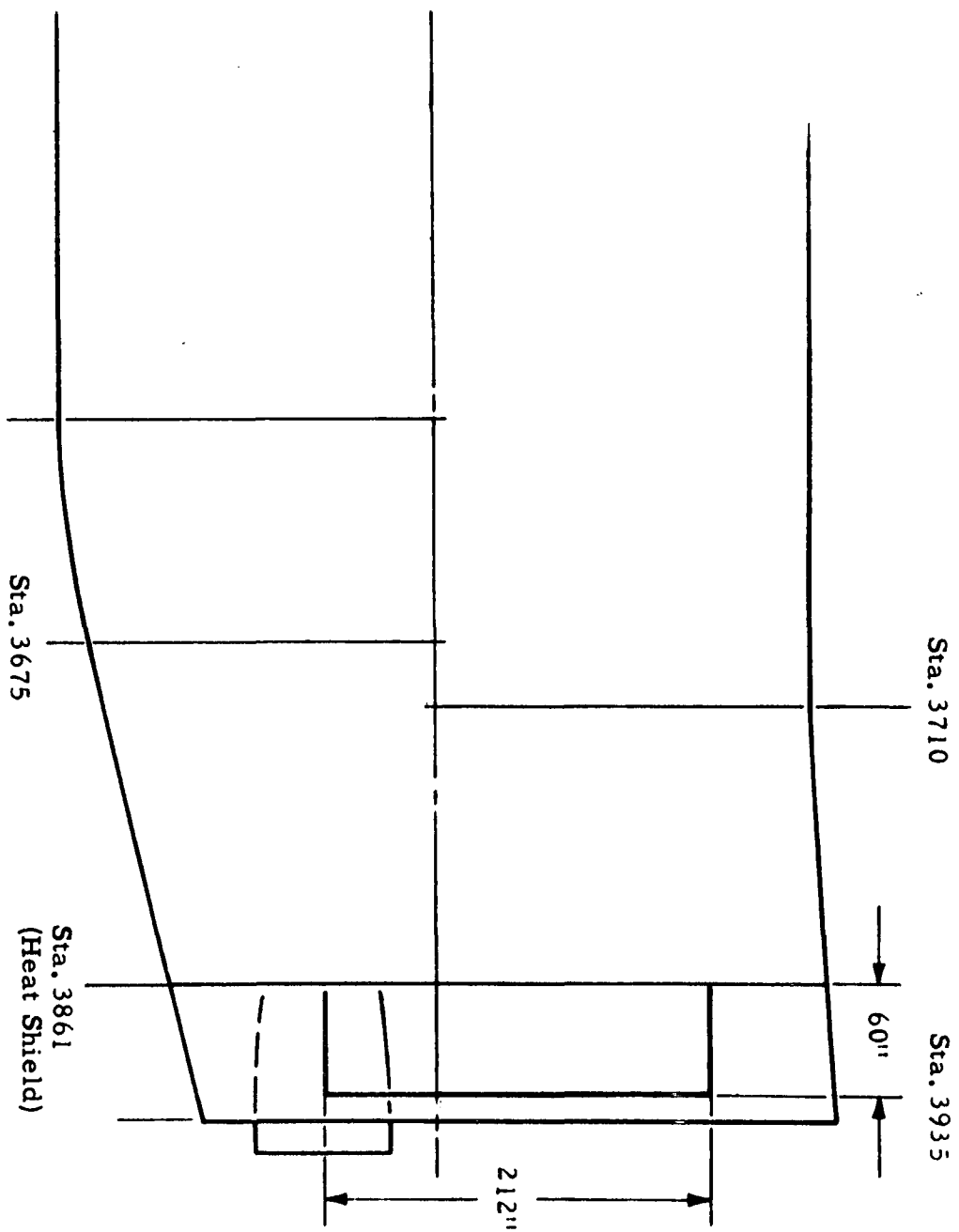


Fig. 8 - Base Venting (Side View)

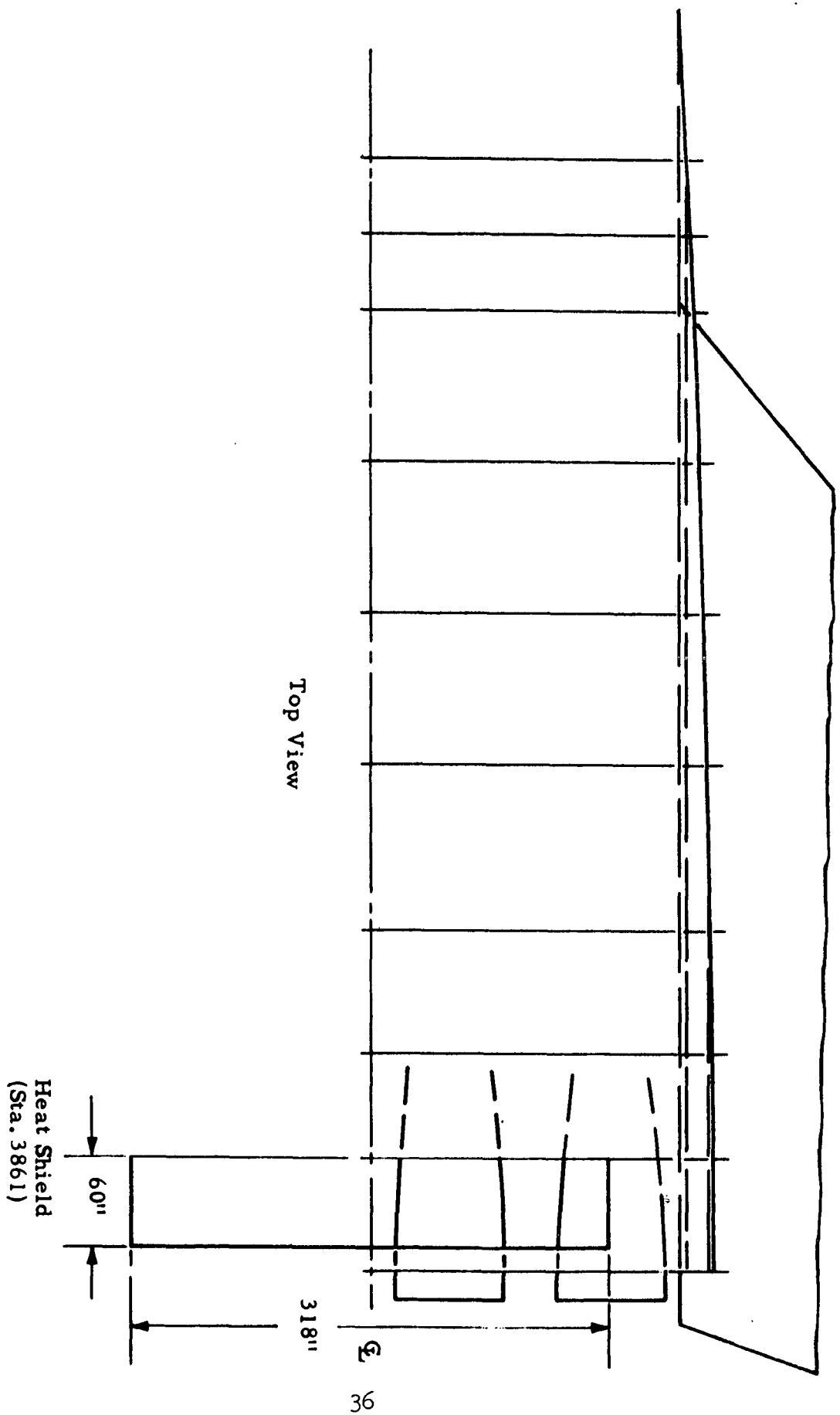


Fig. 9 - Base Venting (Top View)

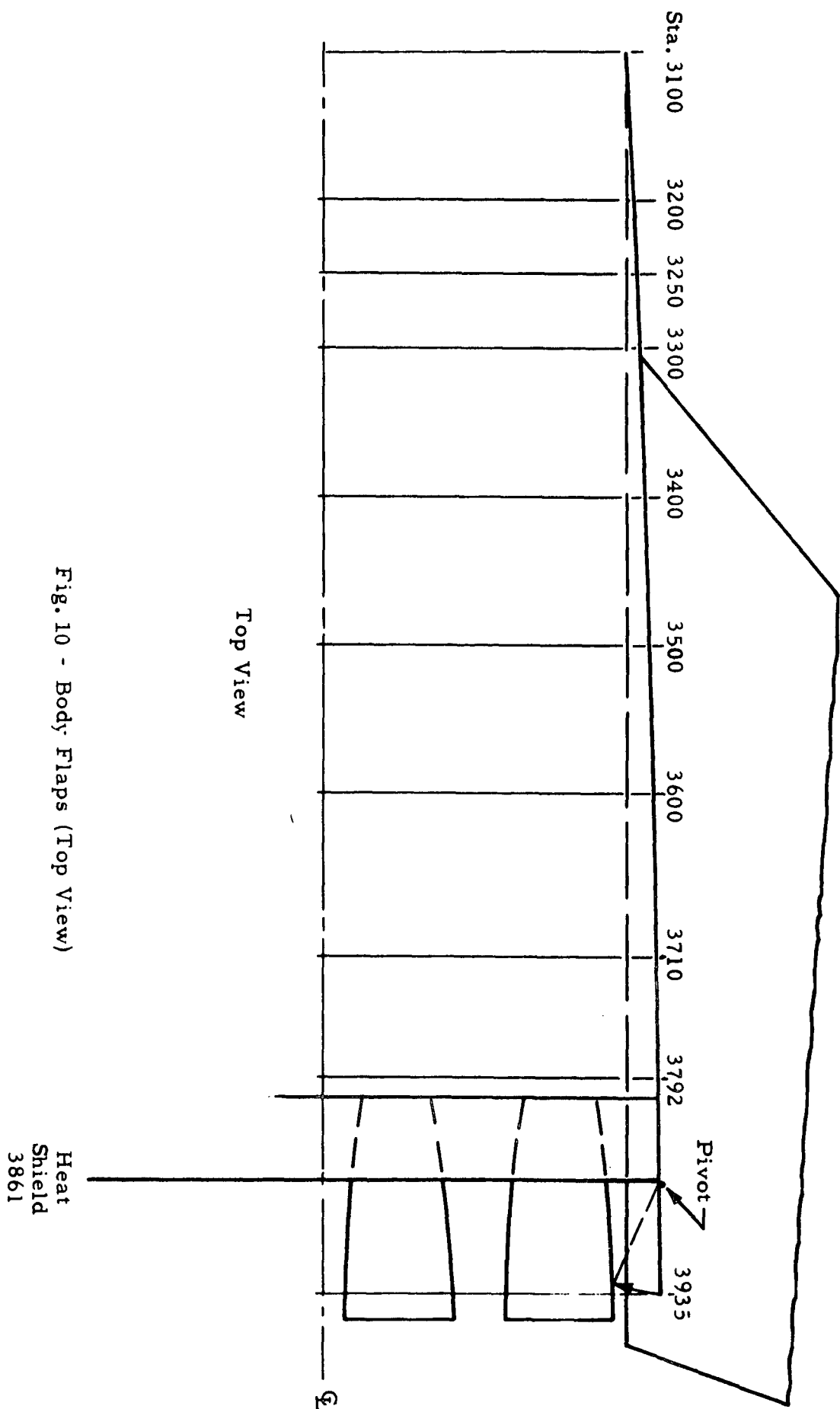


Fig. 10 - Body Flaps (Top View)

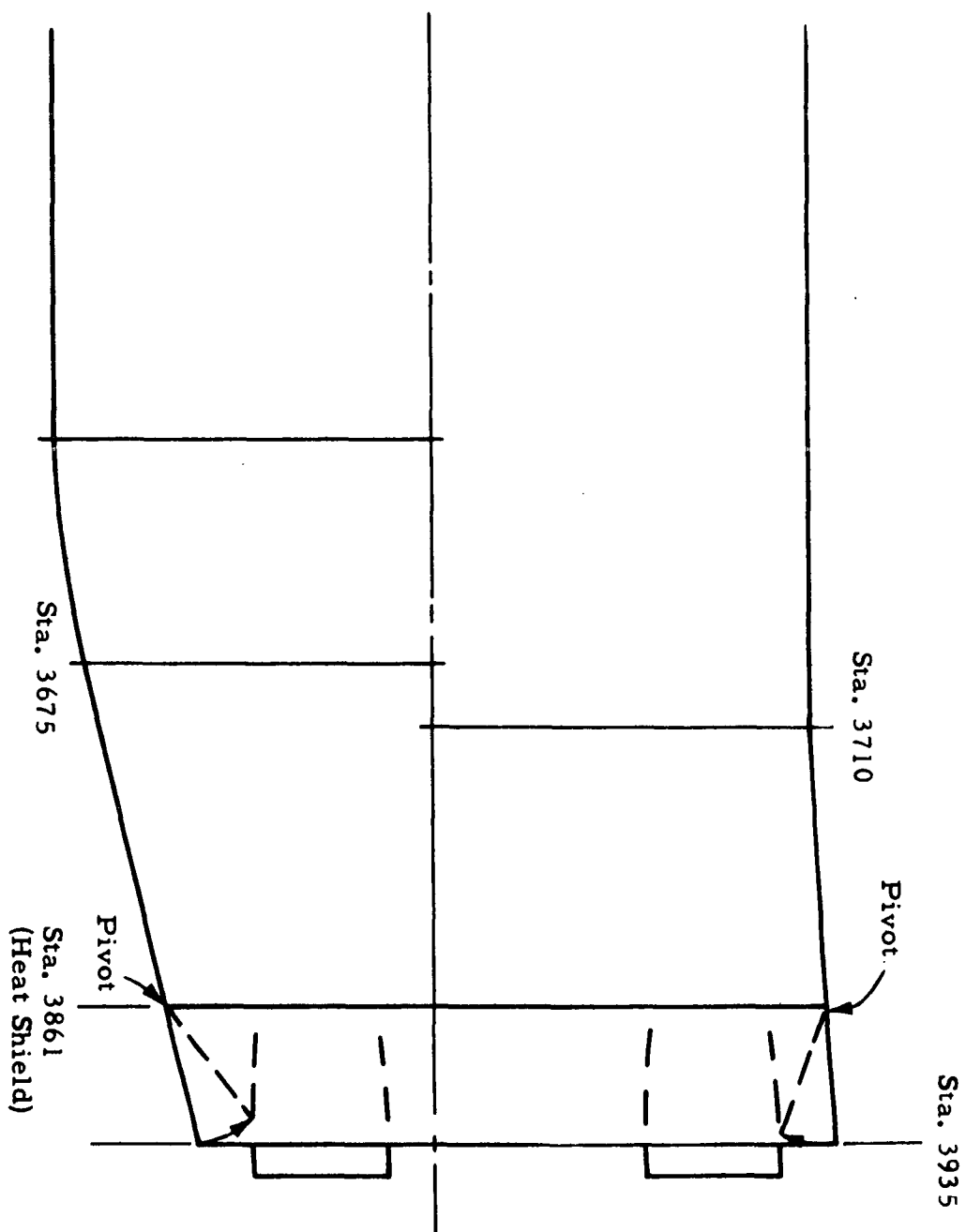


Fig. 11 - Body Flaps (Side View)

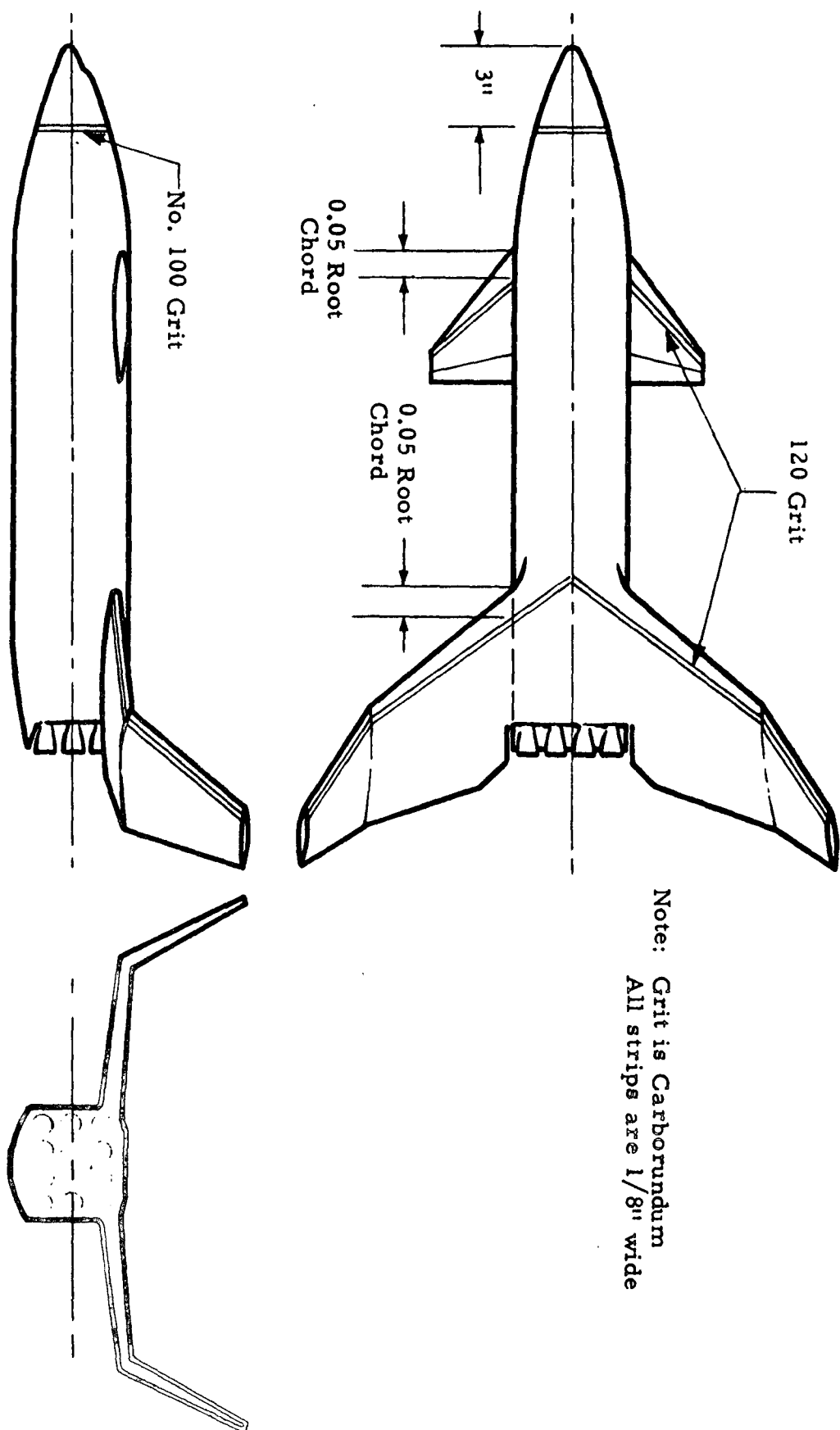
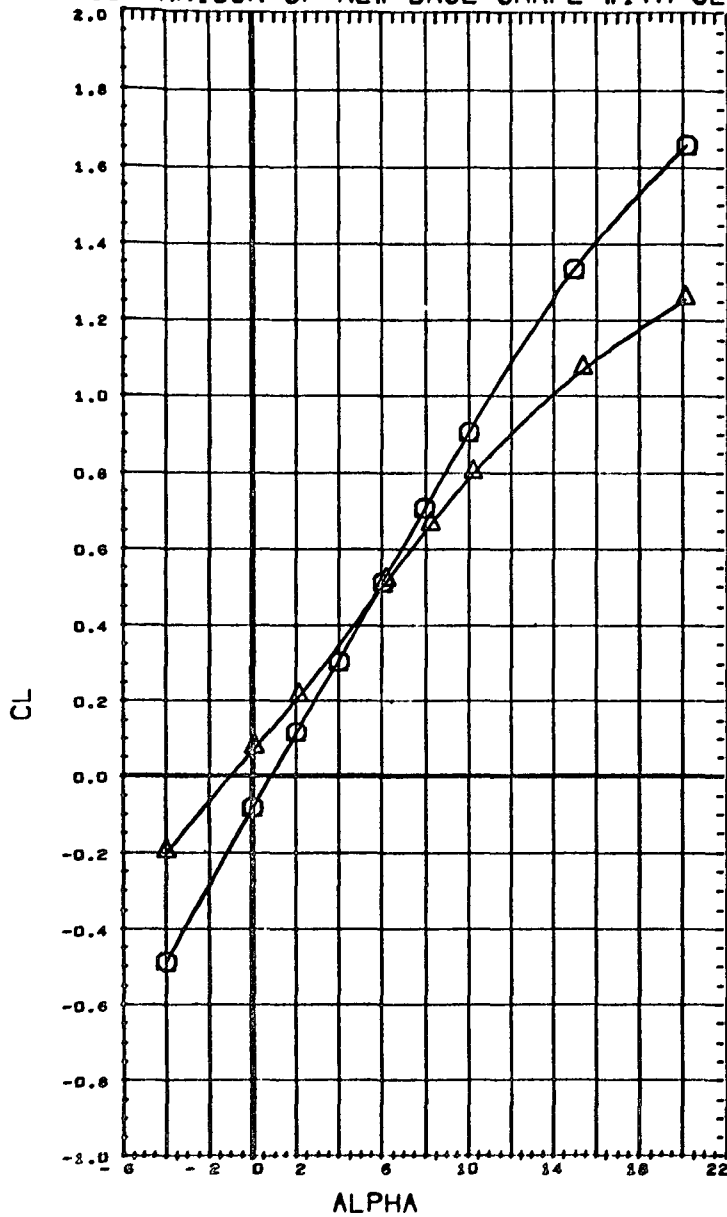


Fig. 12- Typical Transition Grit Installation on Body, Wing and Canard

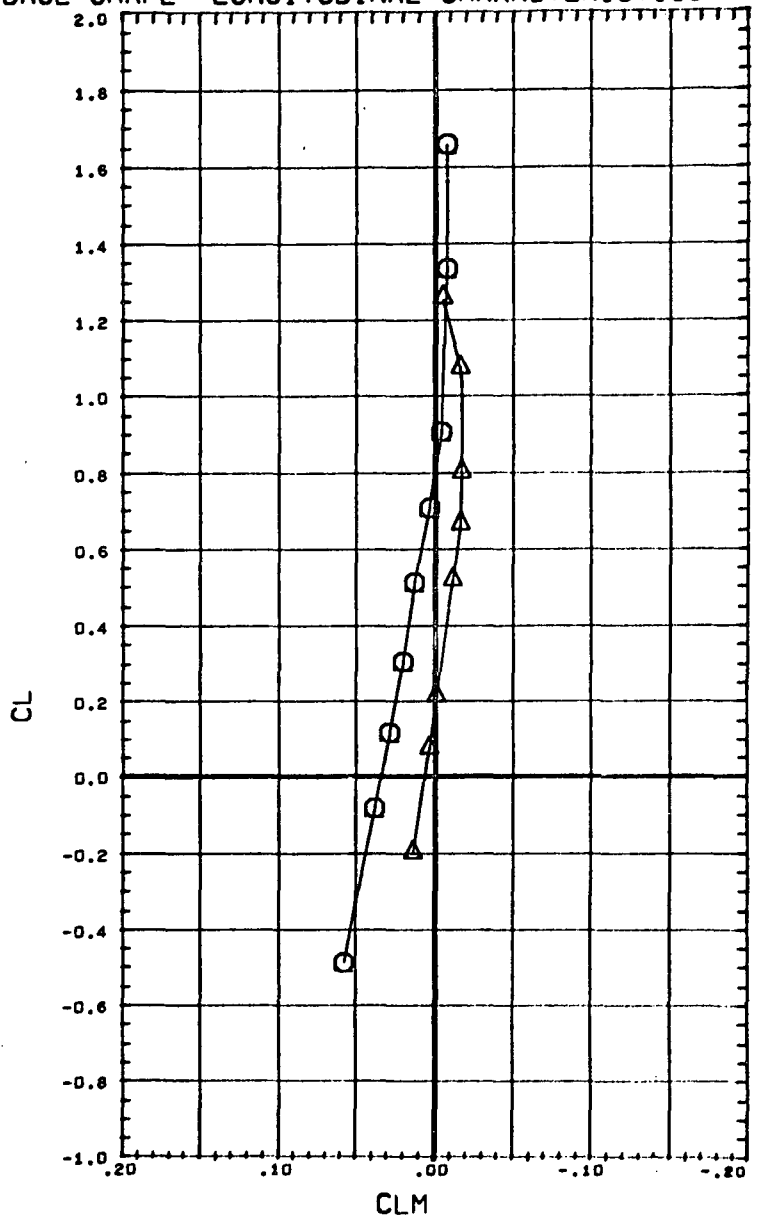
DATA FIGURES

COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



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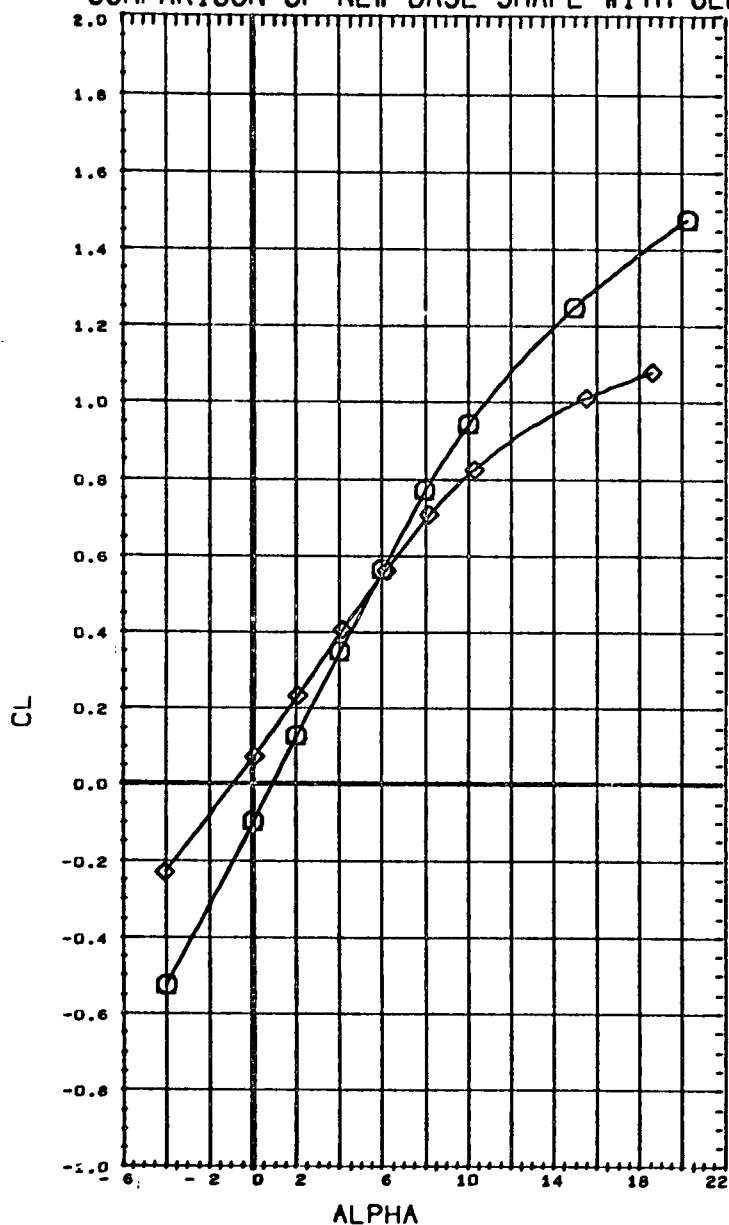
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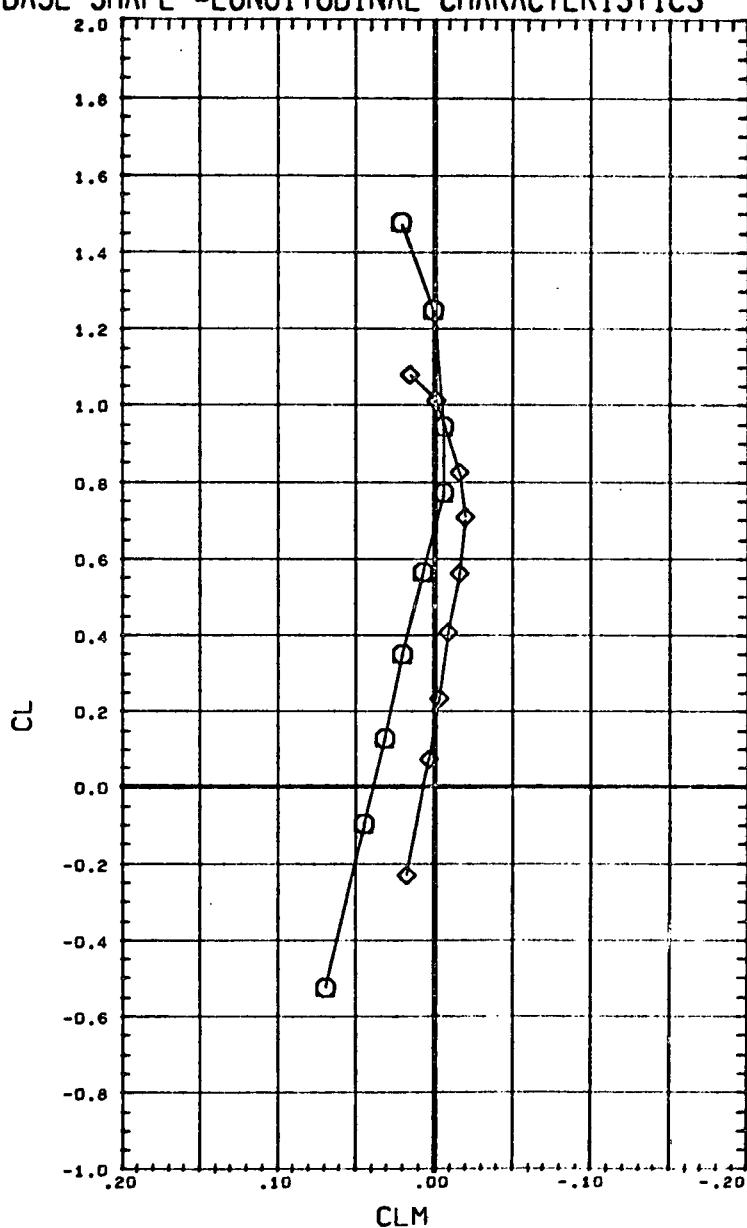
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COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



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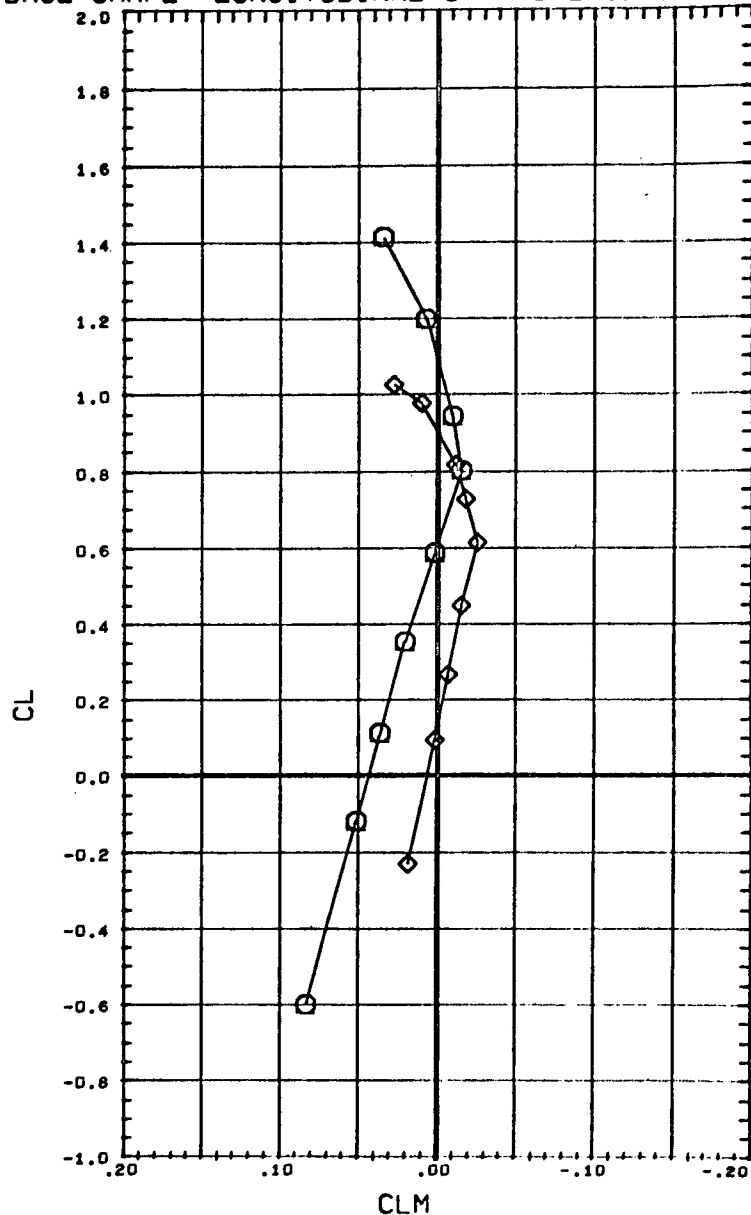
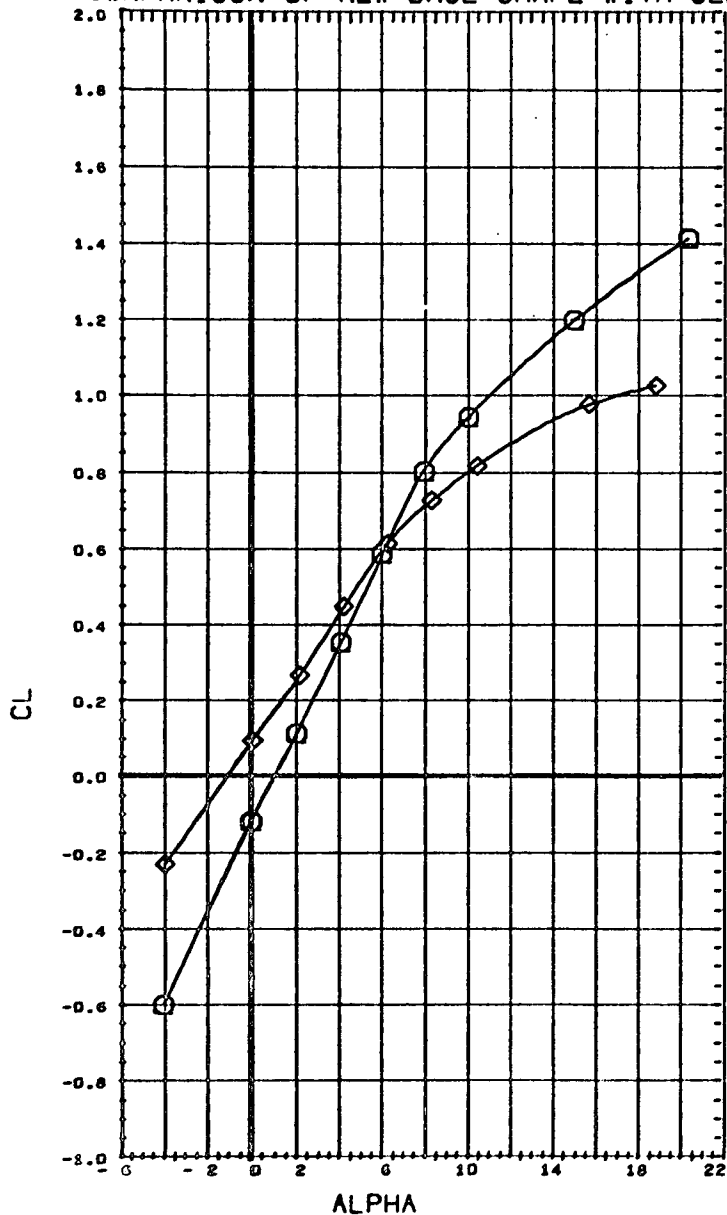
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COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



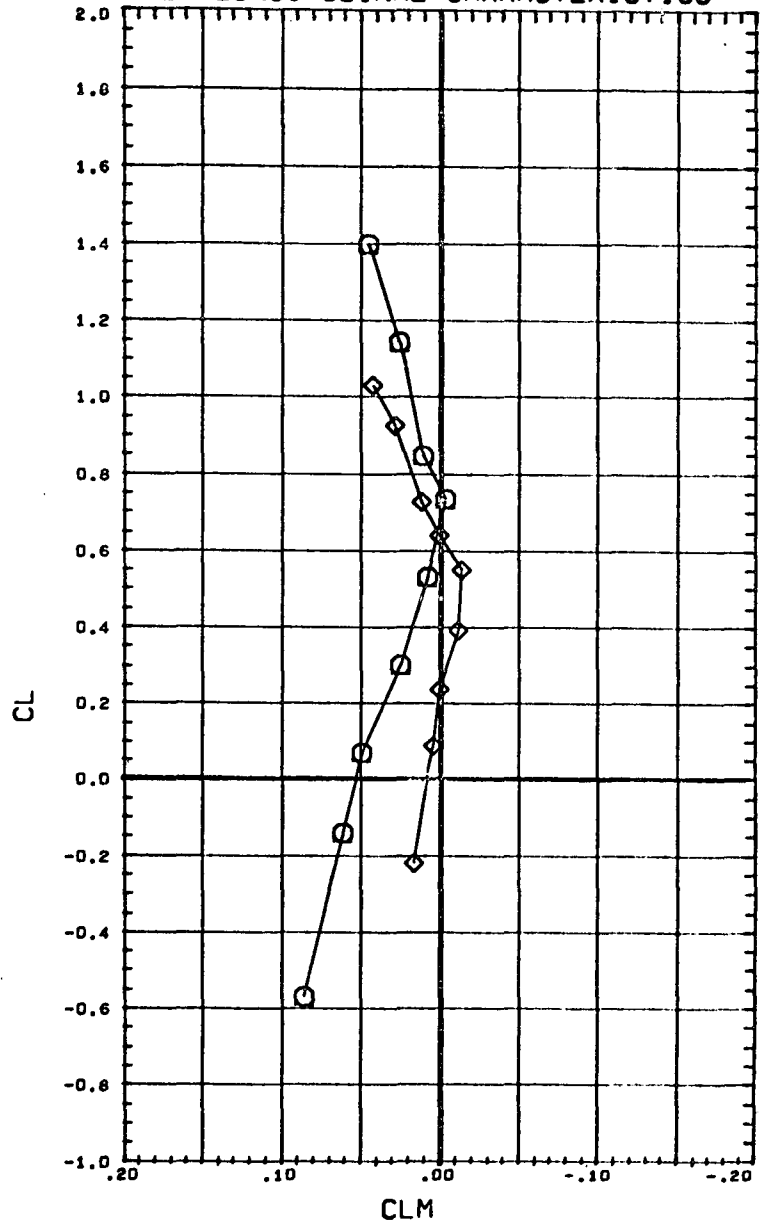
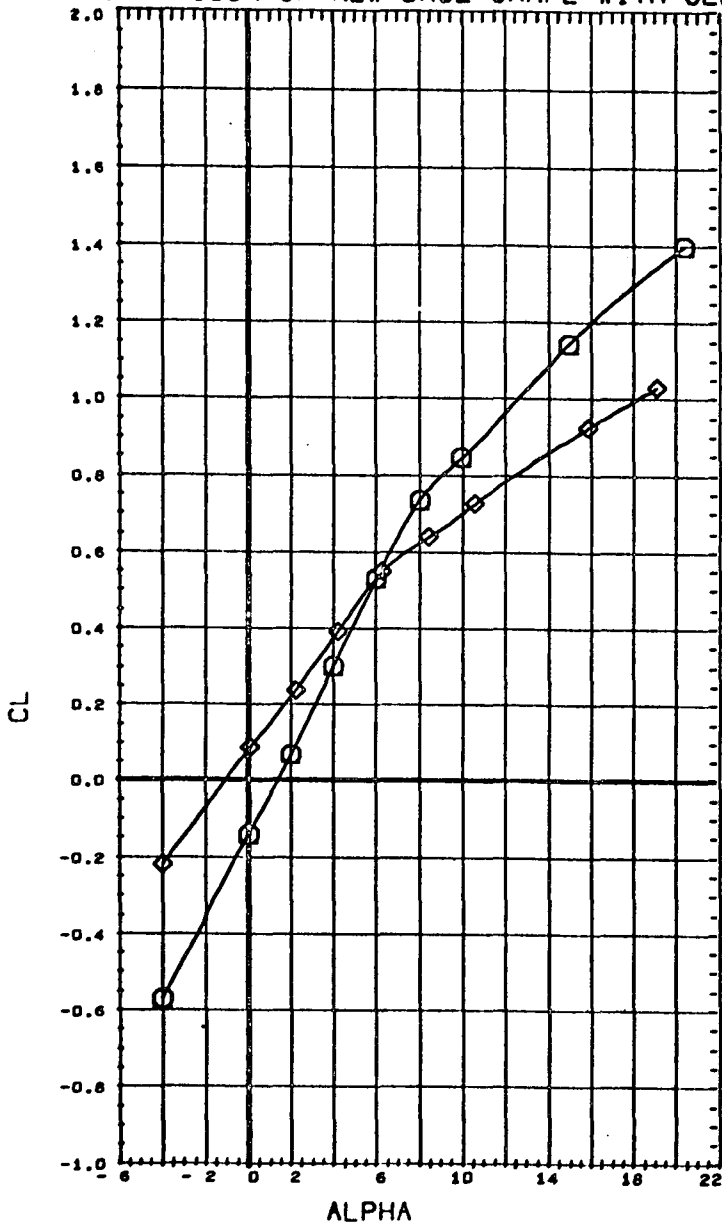
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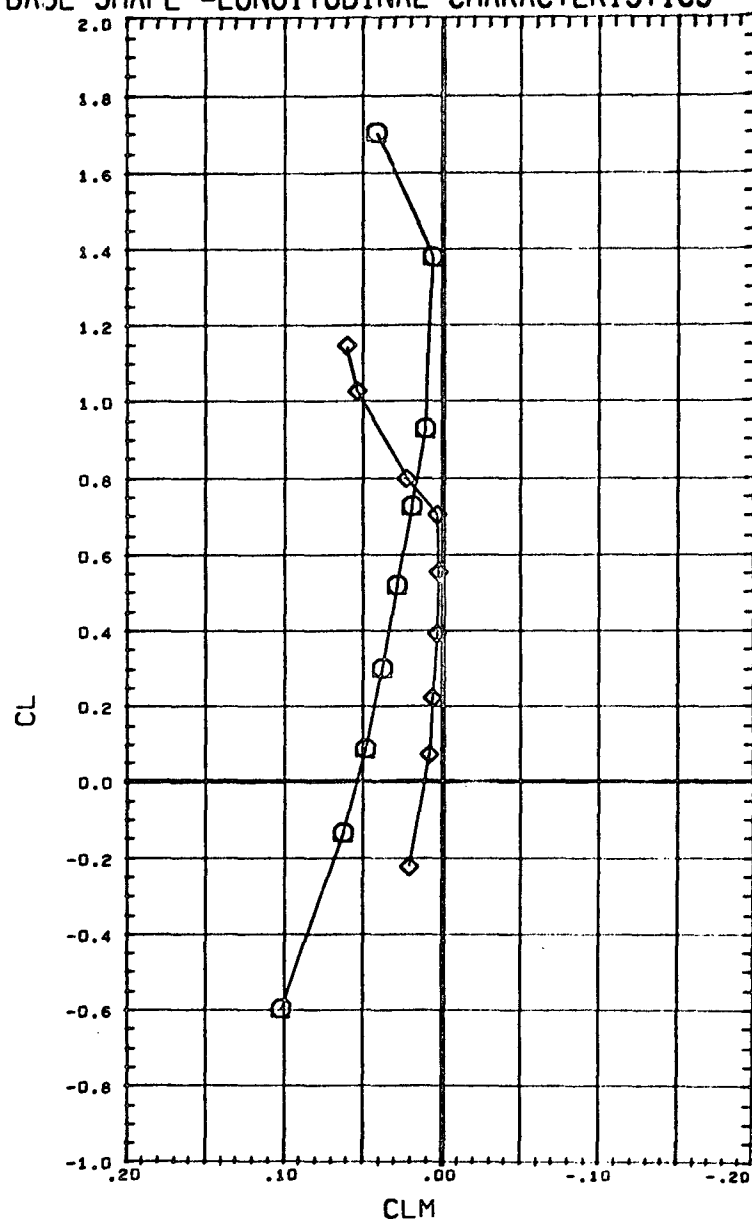
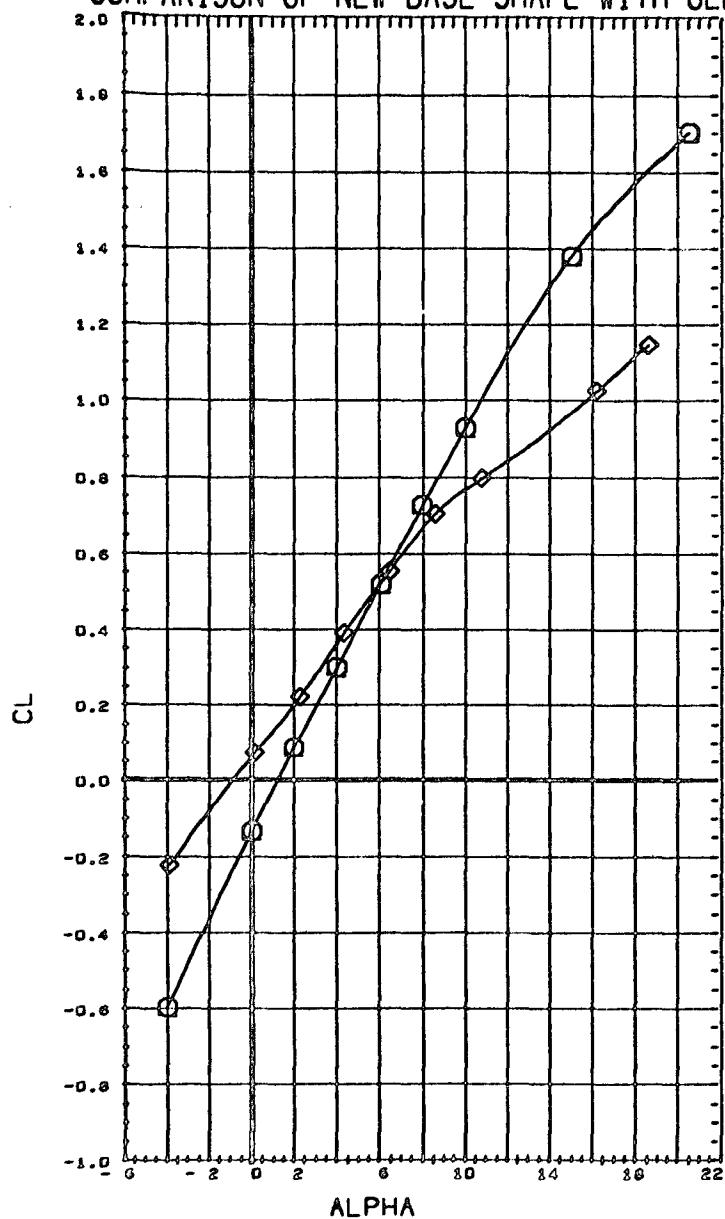
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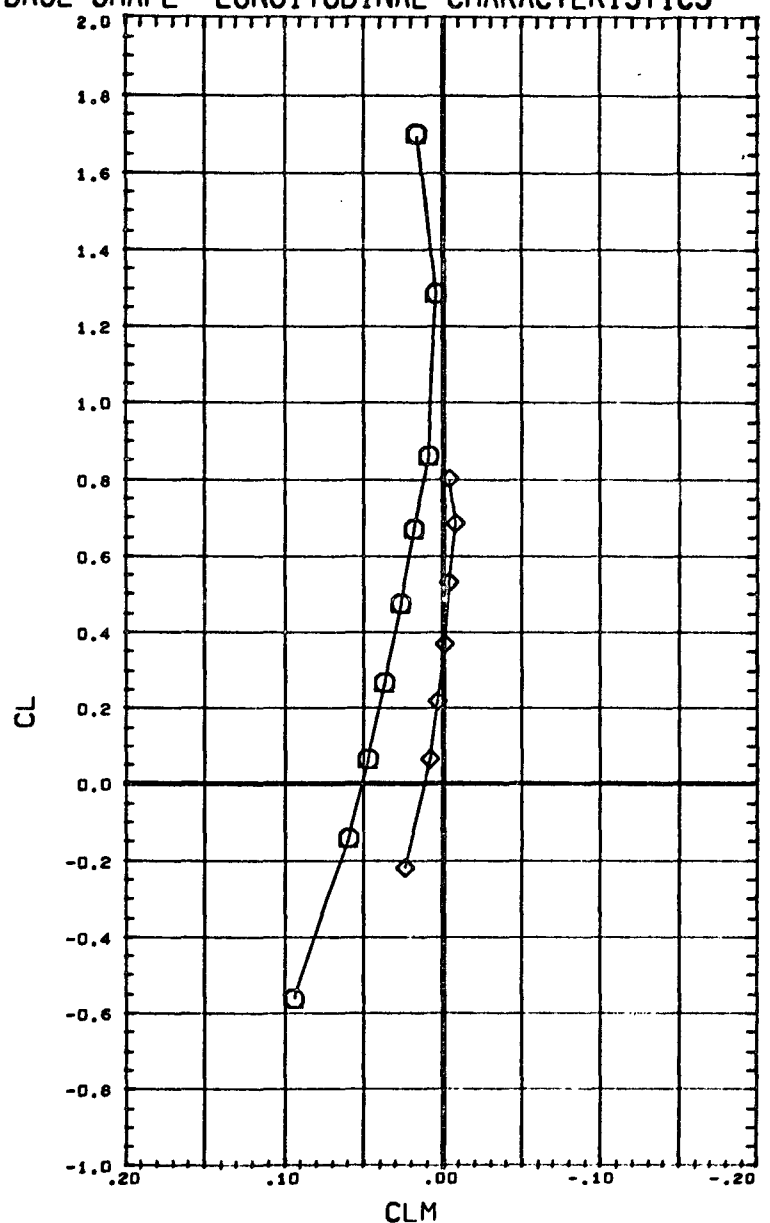
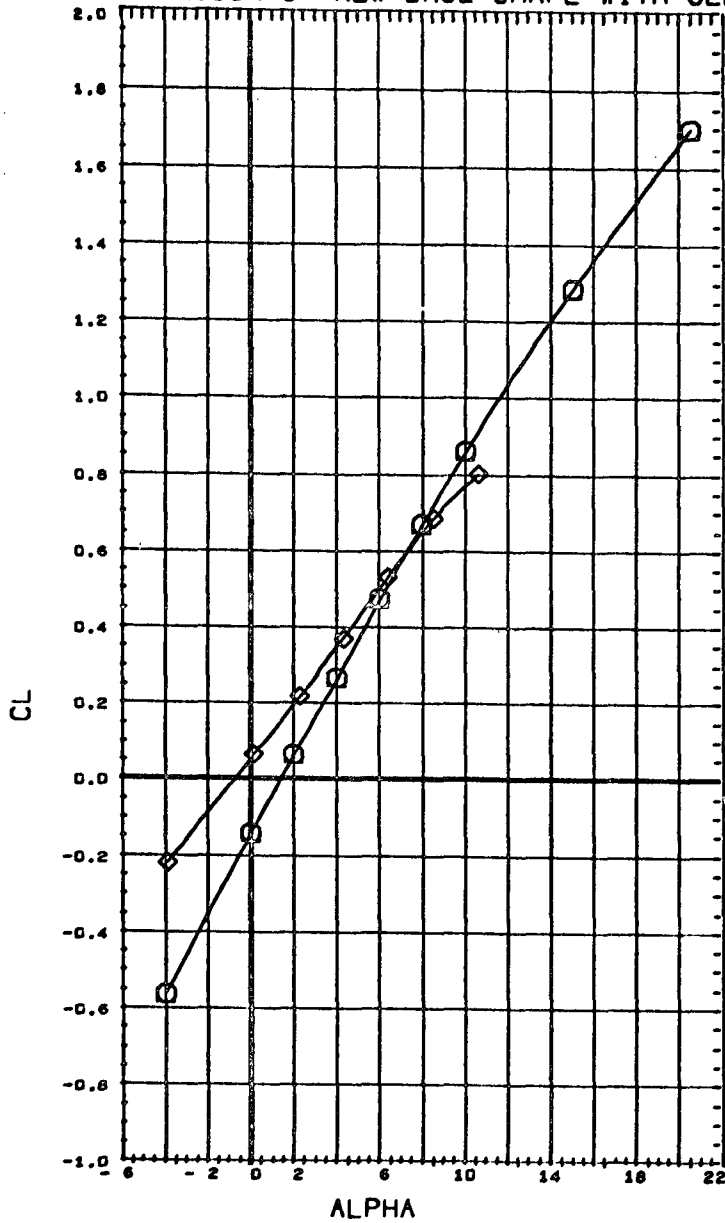
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COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



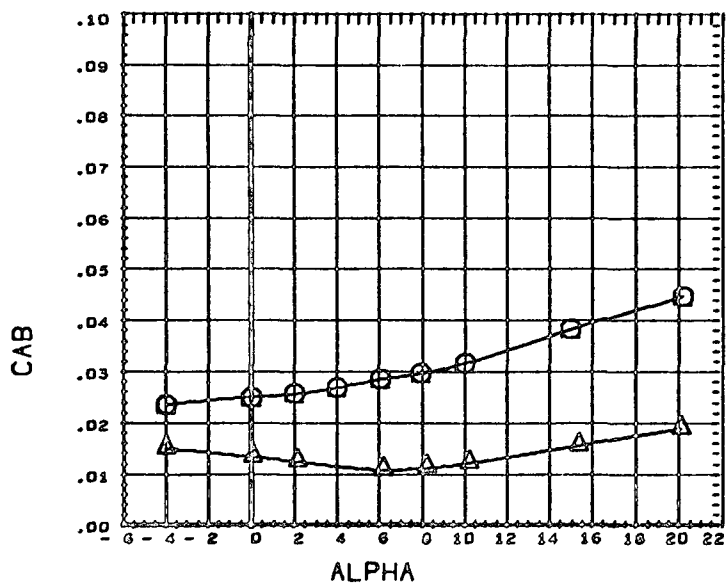
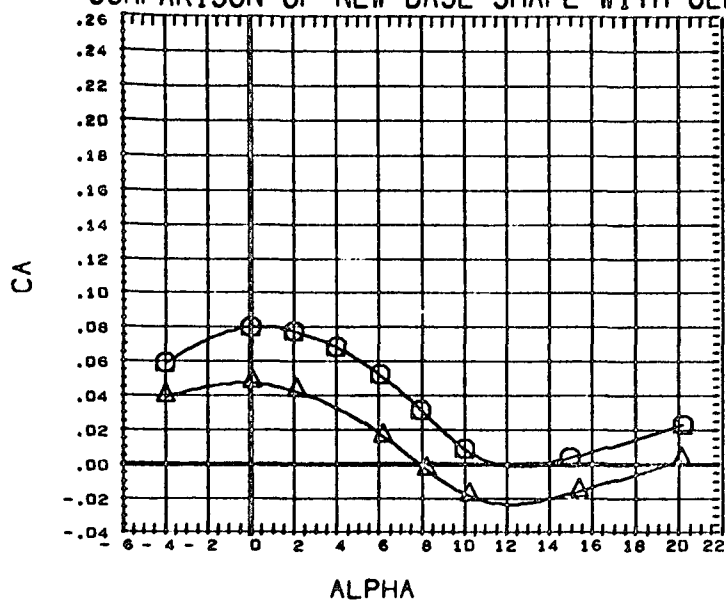
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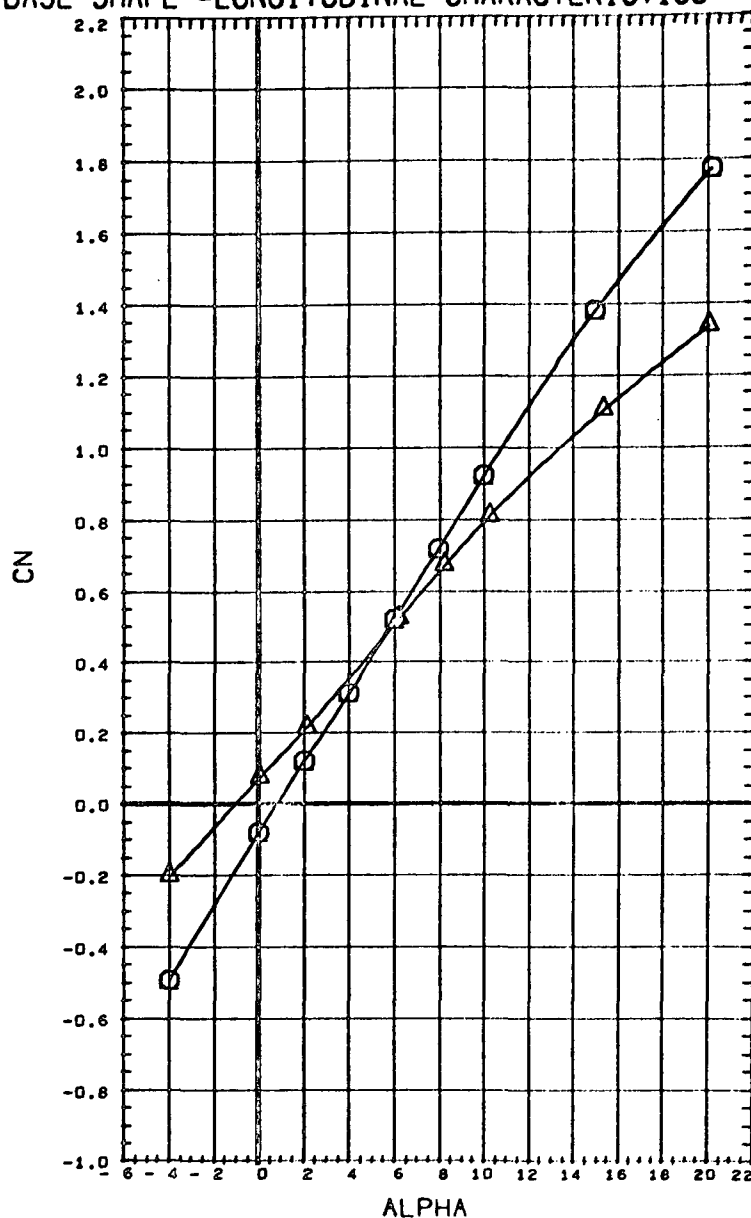
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COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



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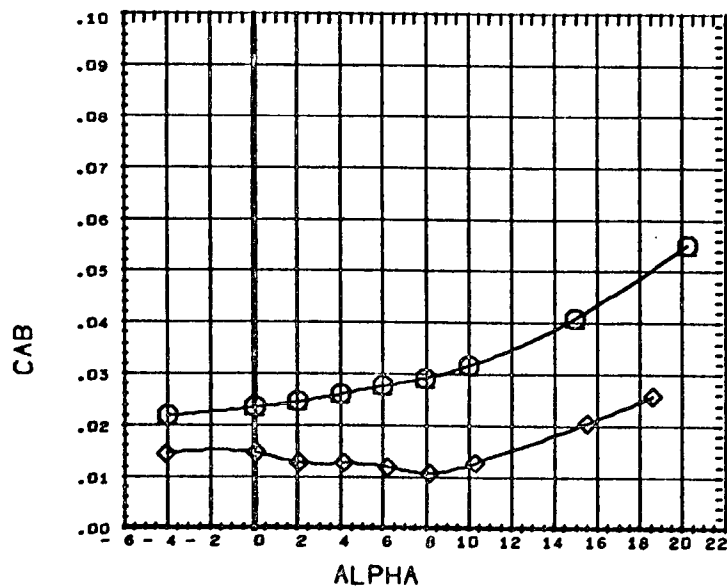
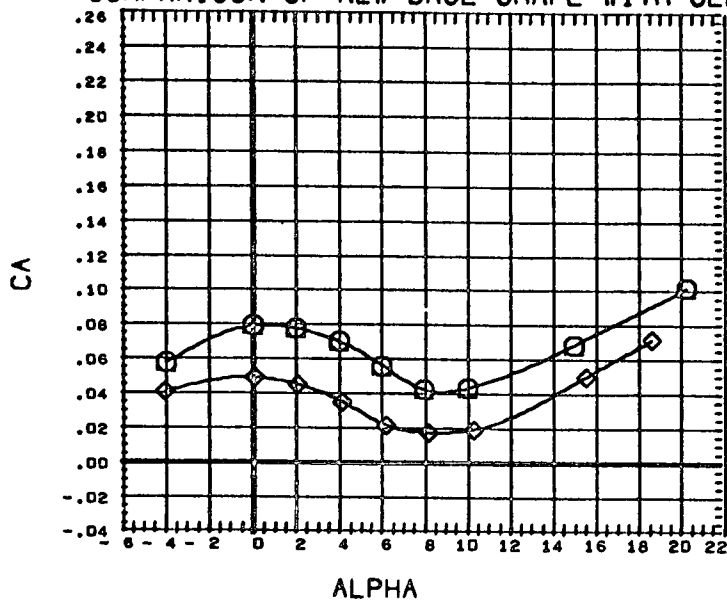
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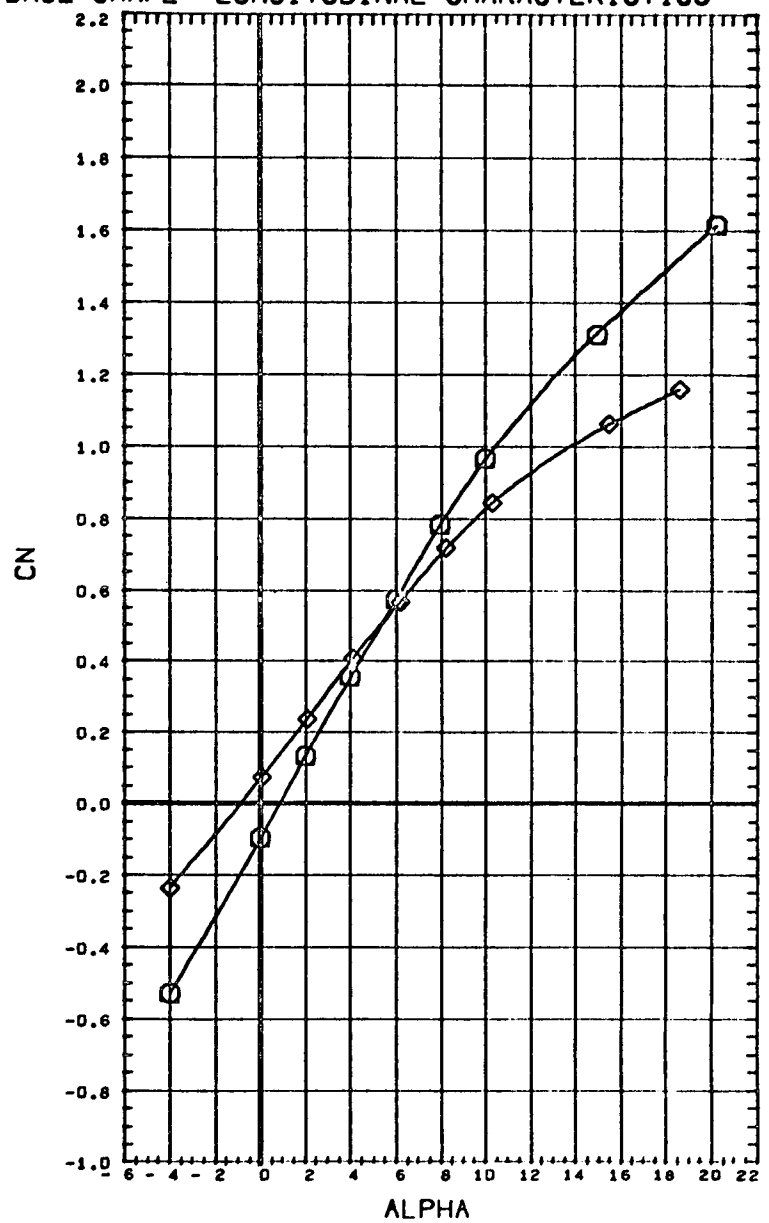
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COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



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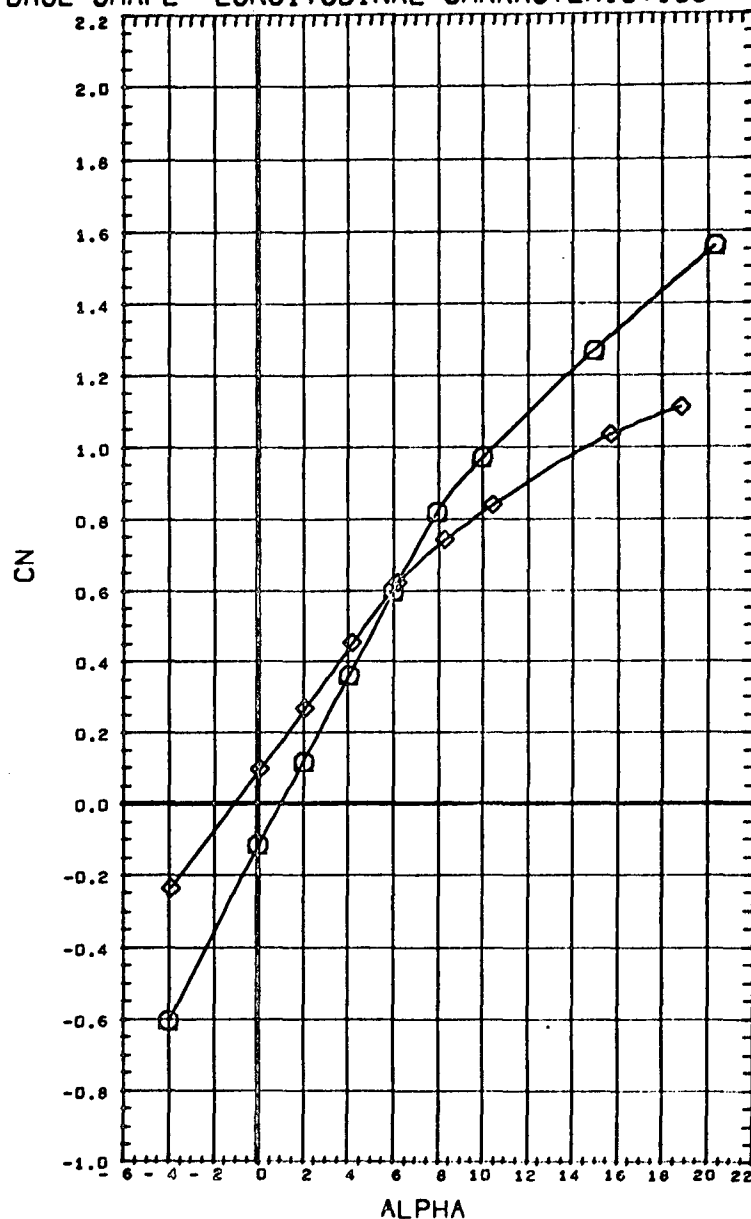
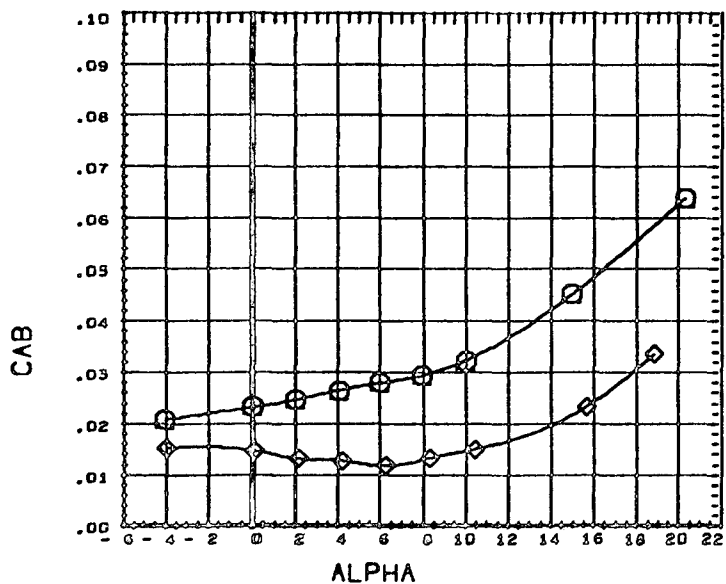
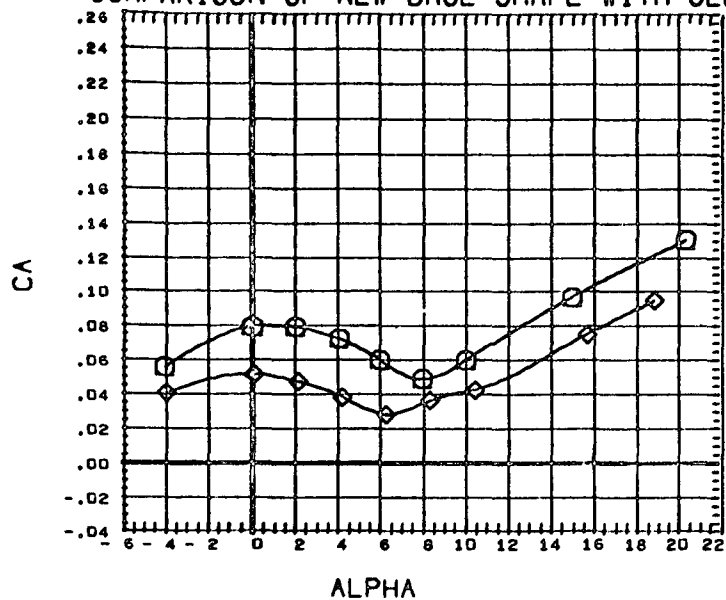
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COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



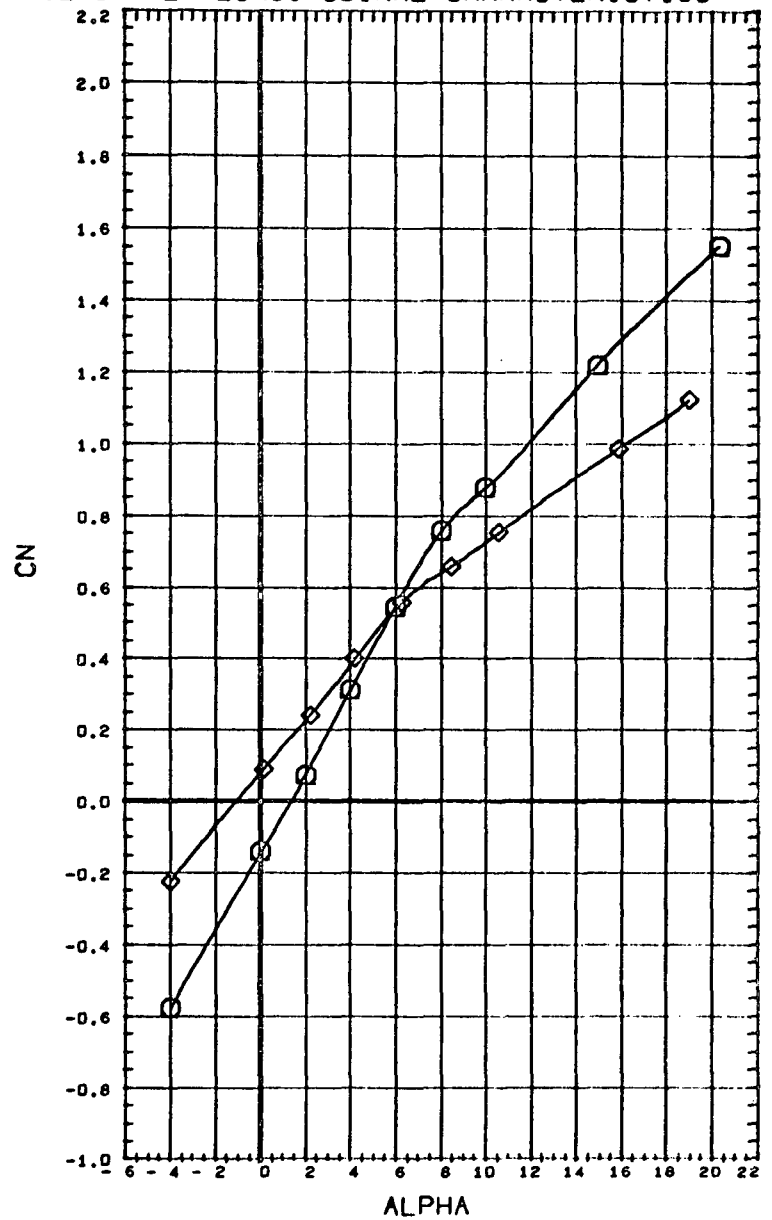
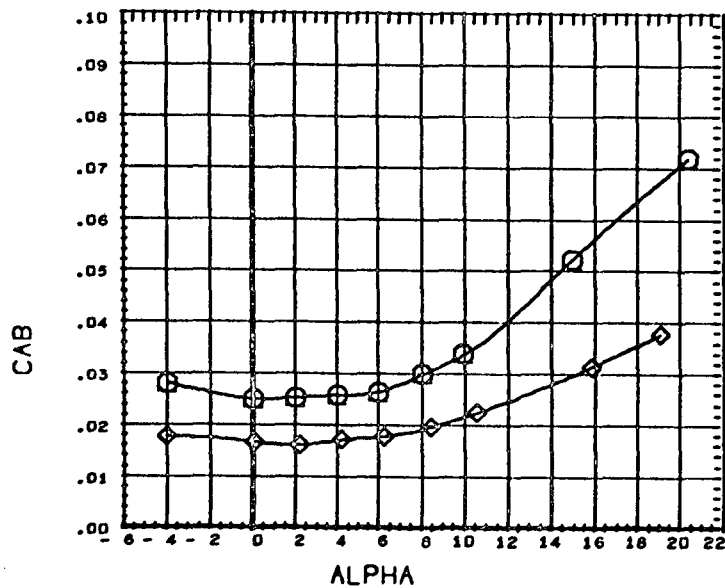
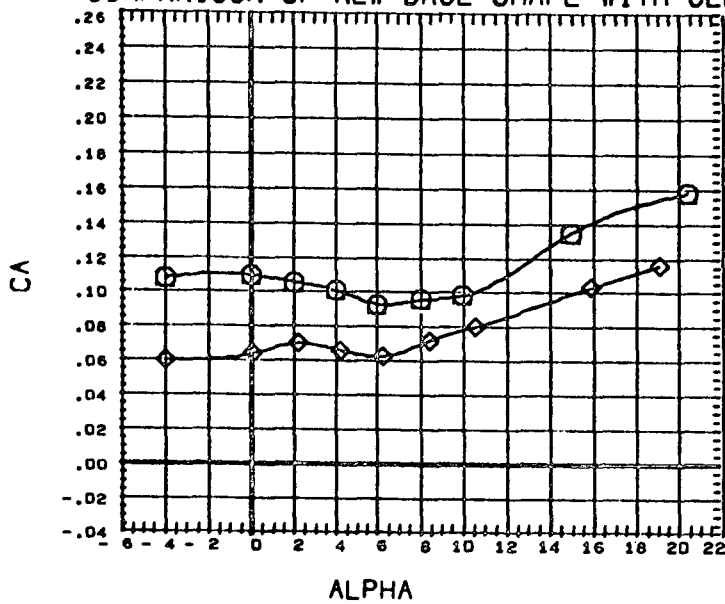
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COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



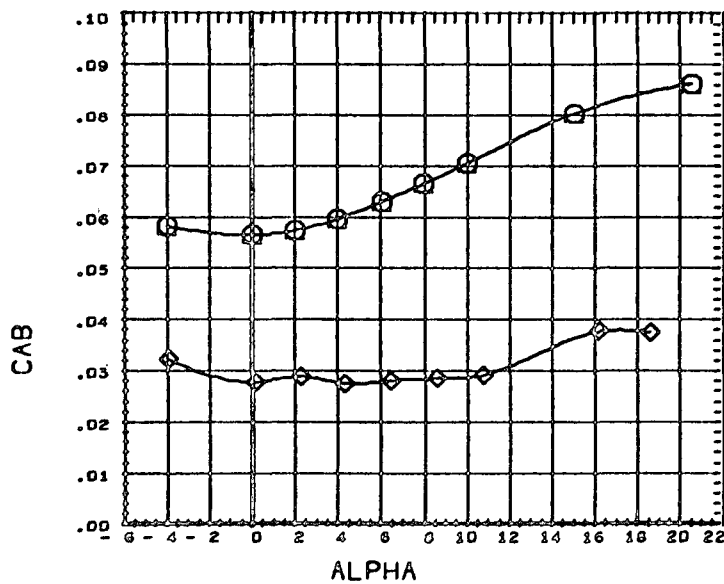
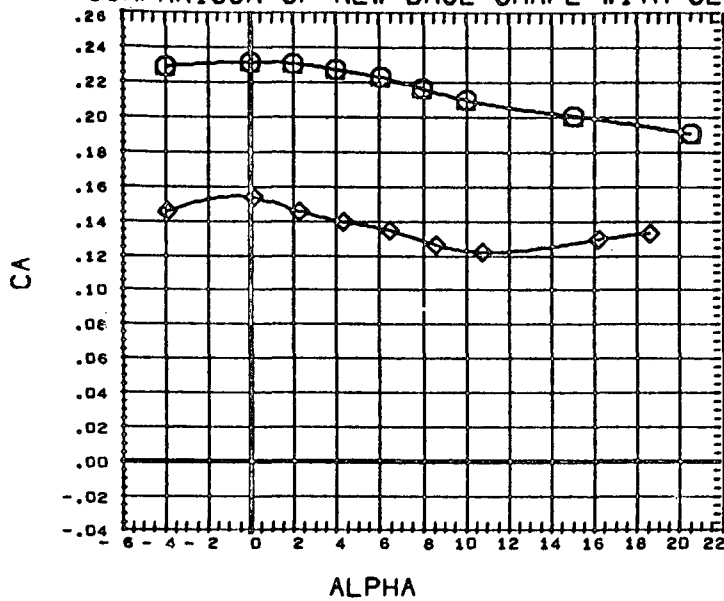
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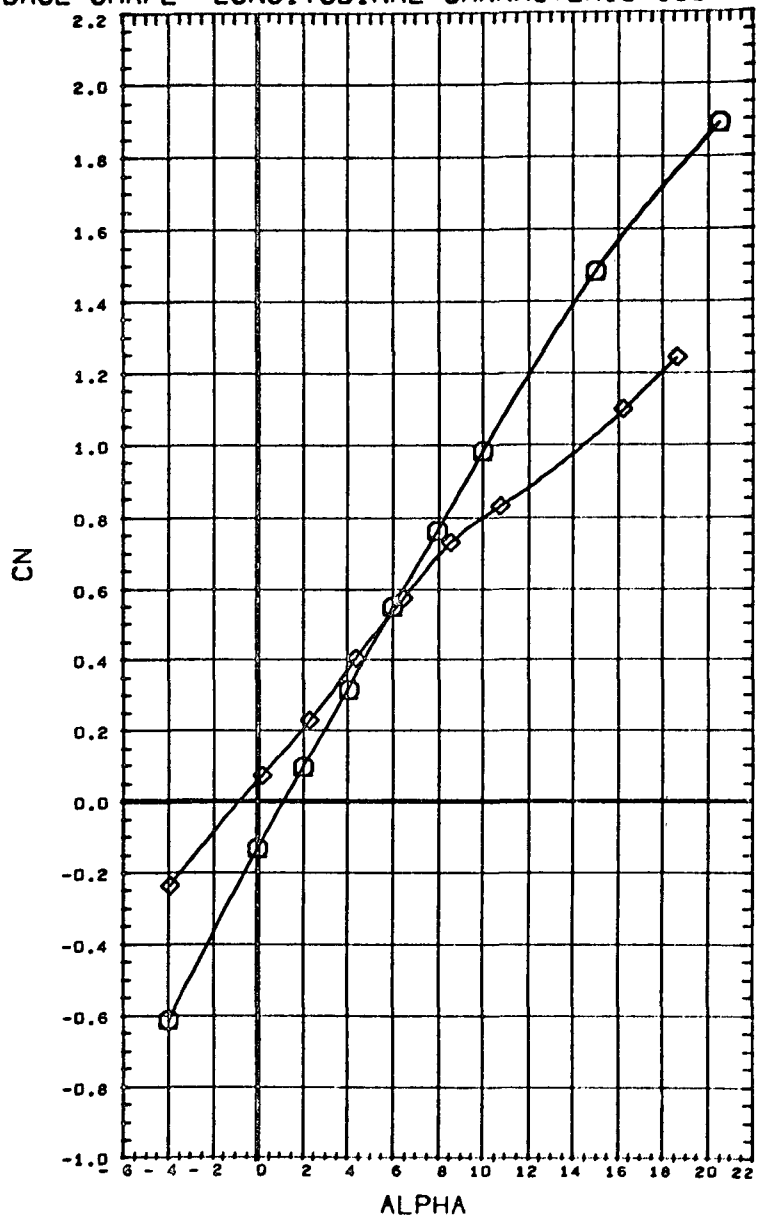
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COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



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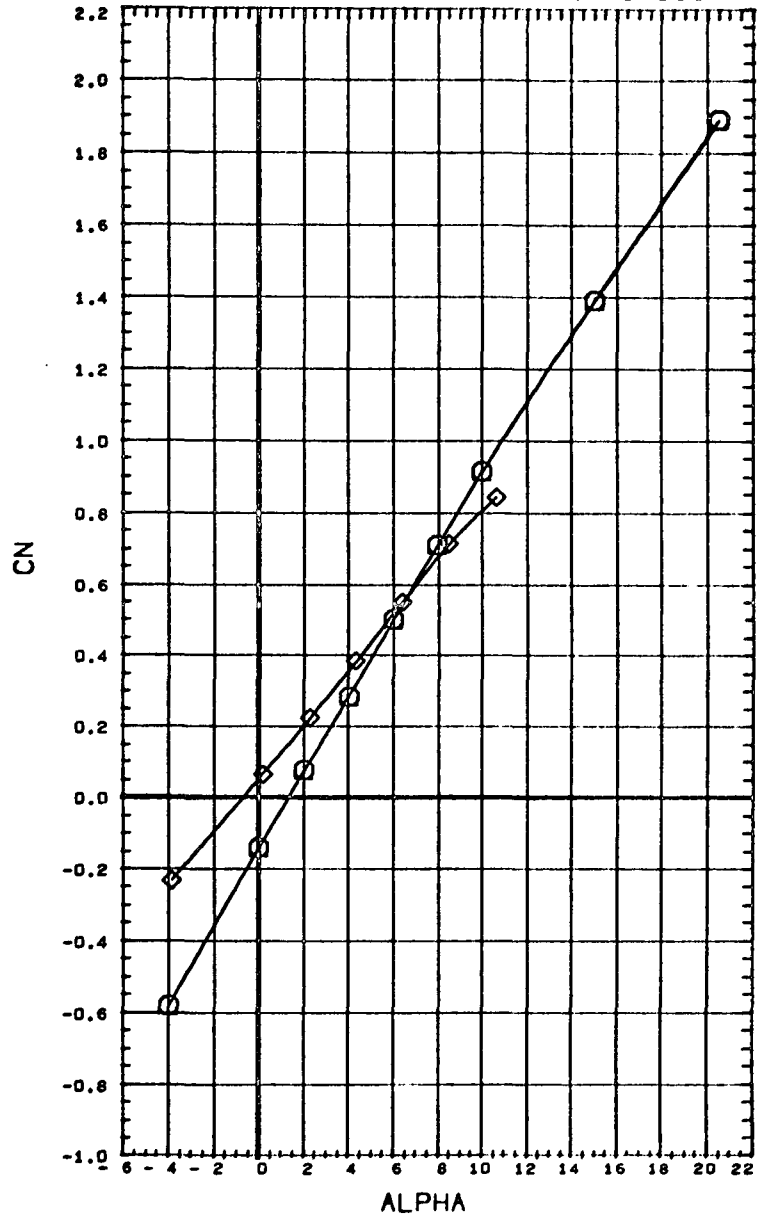
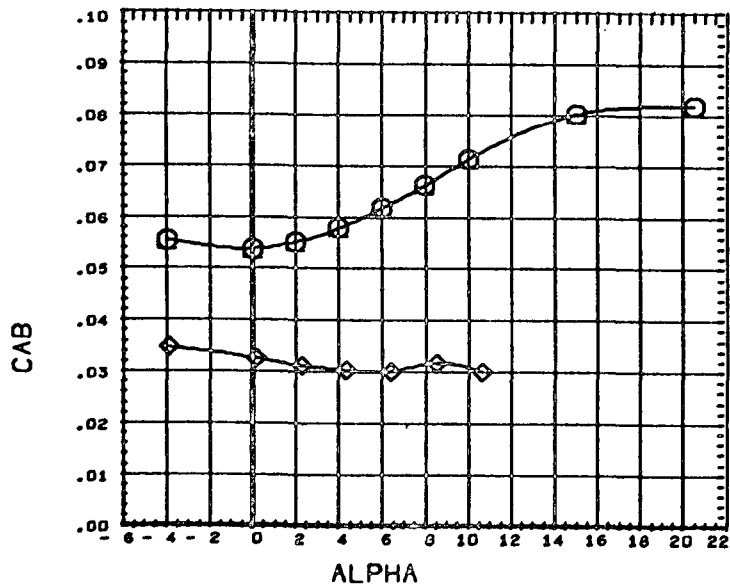
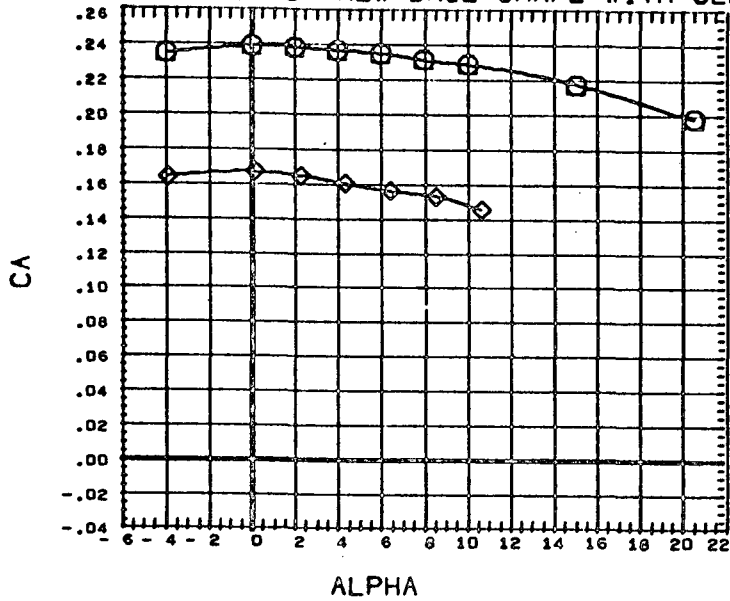
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 SCALE 1.5000 PER CT

COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



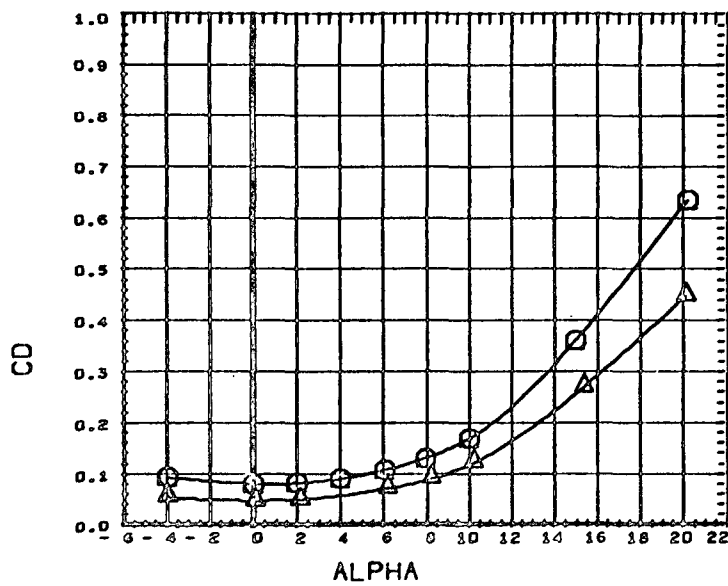
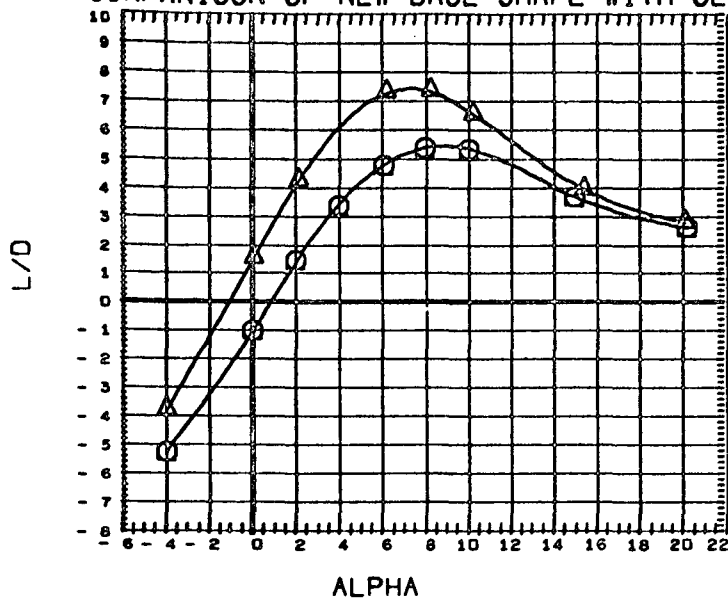
DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RU9001) ○ CAL MSFC/LMSC BOOSTER B4C2F2W3V1
 (AN2001) △ DATA NOT AVAILABLE FOR ALL CONDITIONS
 (AN3001) ◇ NSRDC-3210,MSFC/LMSC BOOSTER B1C2F2W1V1

MACH 1.099

BETA	ELEVTR	CANARD
0.000	0.000	0.000
-0.019	0.000	0.000
0.031	0.000	0.000

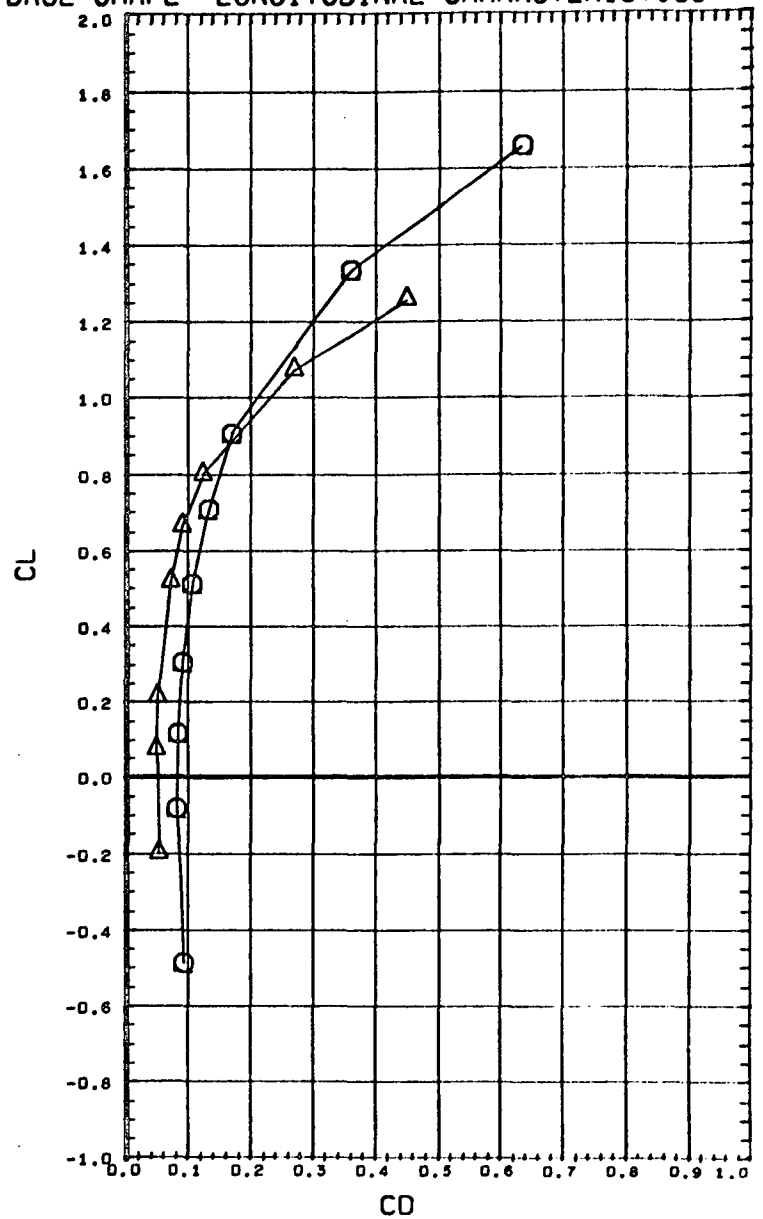
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LREF	3.4530	FT.
BREF	3.4530	FT.
XMRP	2.5950	FT.
YMRP	0.0000	FT.
ZMRP	0.0167	FT.
SCALE	1.5000	PER CT

COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (R09001) ○ CAL MSFC/LMSC BOOSTER B4C2F2W3V1
 (AN2001) △ NSRDC-3110, MSFC/LMSC BOOSTER B1C2F1W1V1
 (AN3001) ◇ DATA NOT AVAILABLE FOR ALL CONDITIONS

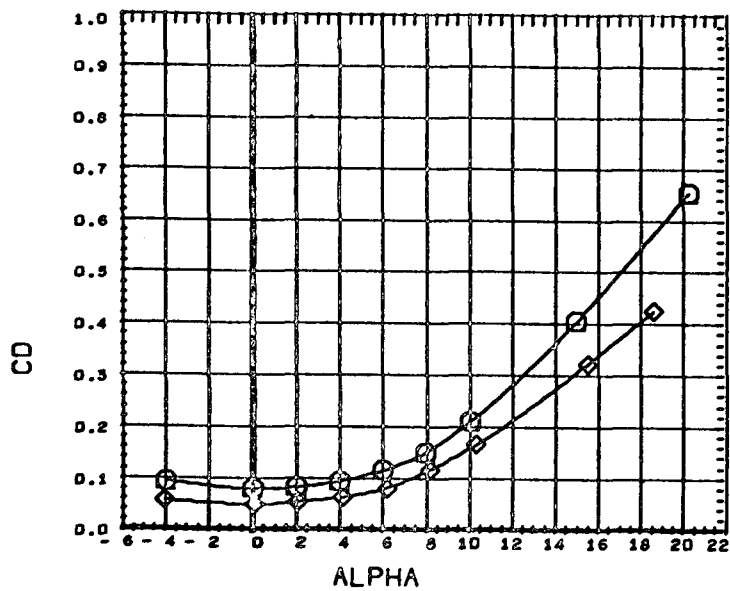
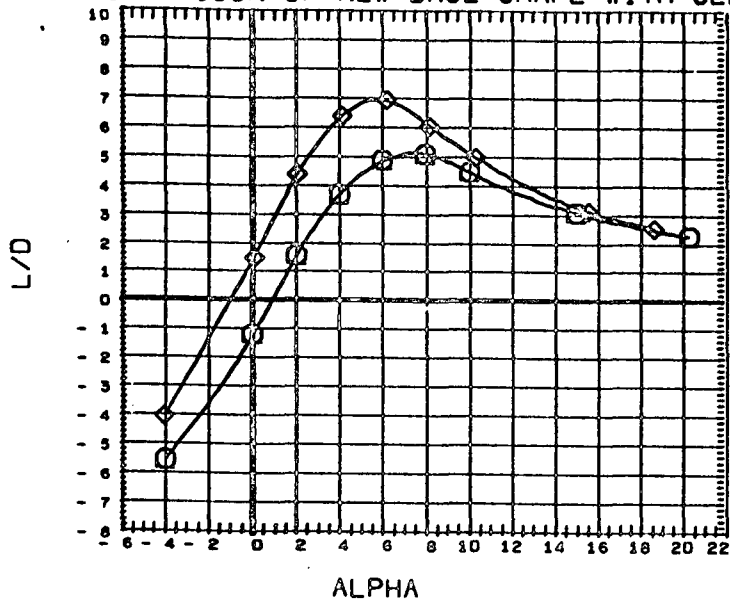
MACH 0.400



BETA	ELEVTR	CANARD
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-0.019	0.000	0.000
0.031	0.000	0.000

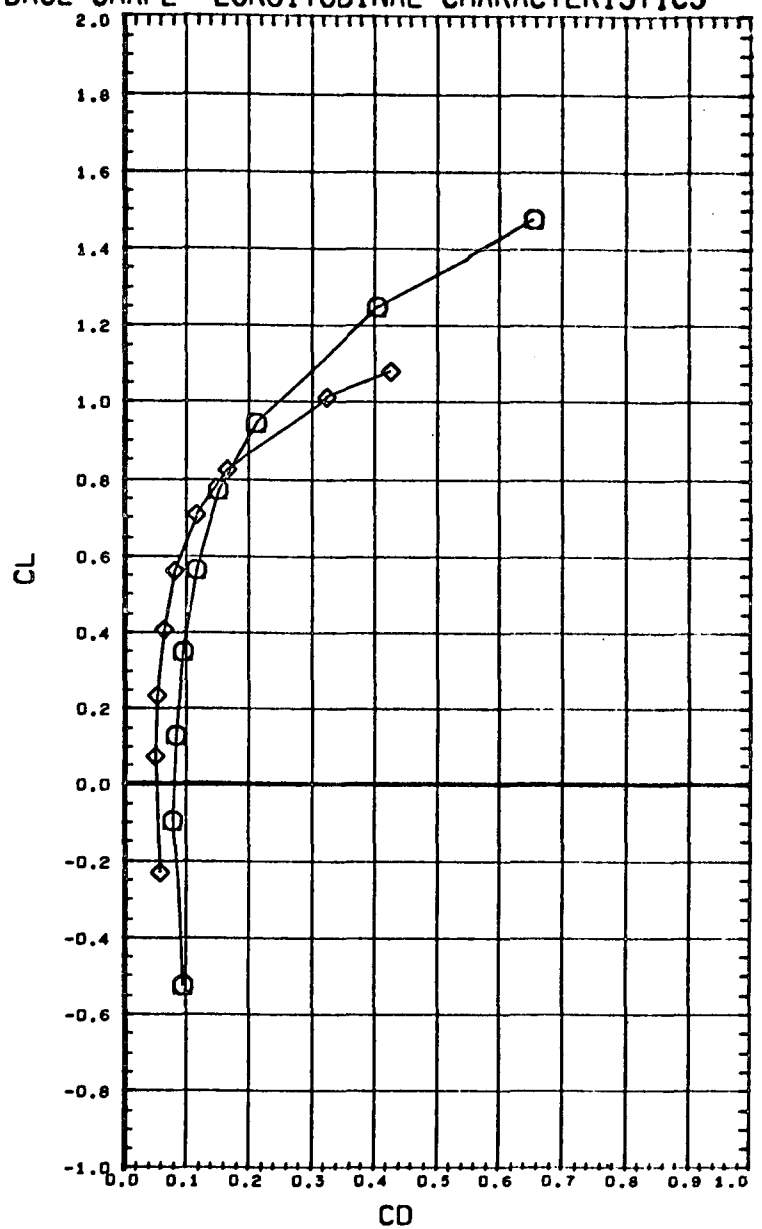
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 LREF 3.4530 FT.
 BREF 3.4530 FT.
 XMRP 2.5950 FT.
 YMRP 0.0000 FT.
 ZMRP 0.0167 FT.
 SCALE 1.5000 PER CT

COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RU9001) CAL MSFC/LMSC BOOSTER B4C2F2W3V1
 (AN2001) DATA NOT AVAILABLE FOR ALL CONDITIONS
 (AN3001) NSRDC-3210,MSFC/LMSC BOOSTER B1C2F2W1V1

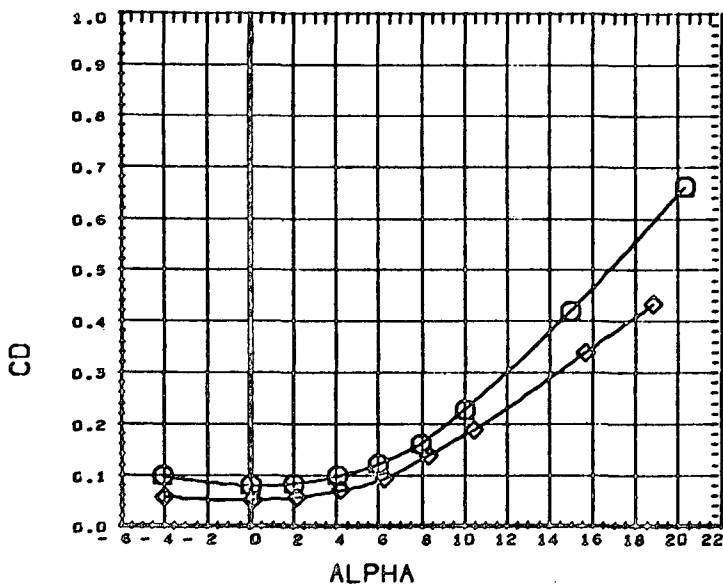
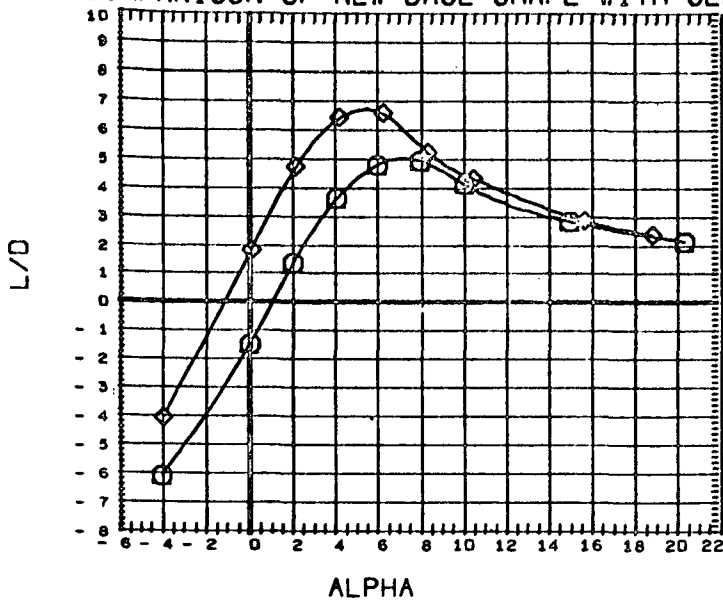
MACH 0.698



BETA ELEVTR CANARD
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 -0.019 0.000 0.000
 0.031 0.000 0.000

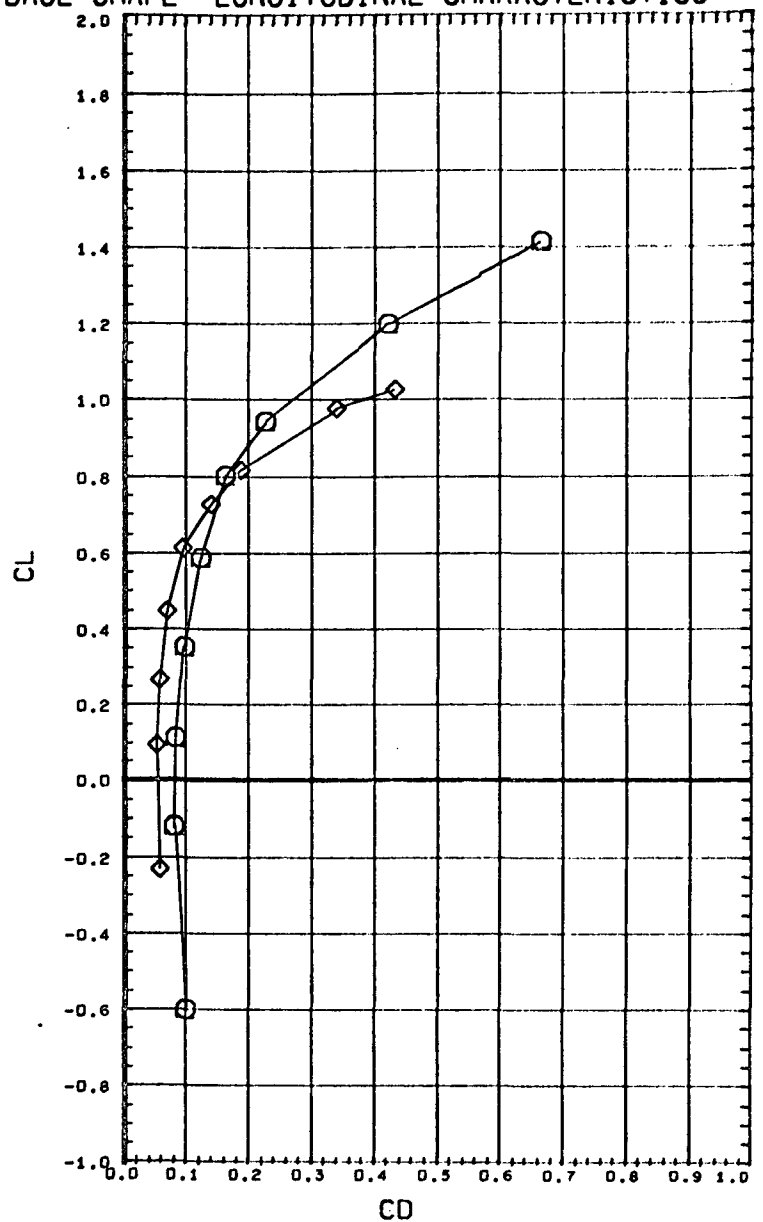
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 LREF 3.4530 FT.
 BREF 3.4530 FT.
 XMRP 2.5950 FT.
 YMRP 0.0000 FT.
 ZMRP 0.0187 FT.
 SCALE 1.5000 PER CT

COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RU9001) CAL MSFC/LMSC BOOSTER B4C2F2W3V1
 (AN2001) DATA NOT AVAILABLE FOR ALL CONDITIONS
 (AN3001) NSRDC-3210, MSFC/LMSC BOOSTER B1C2F2W1V1

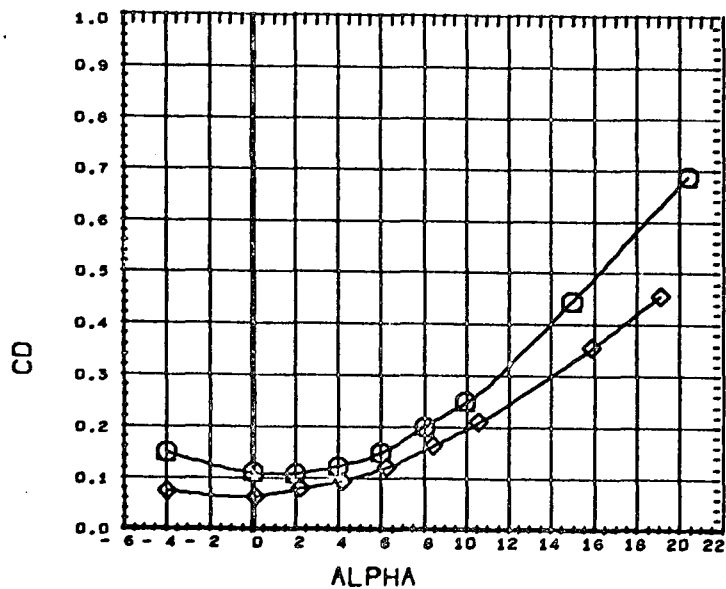
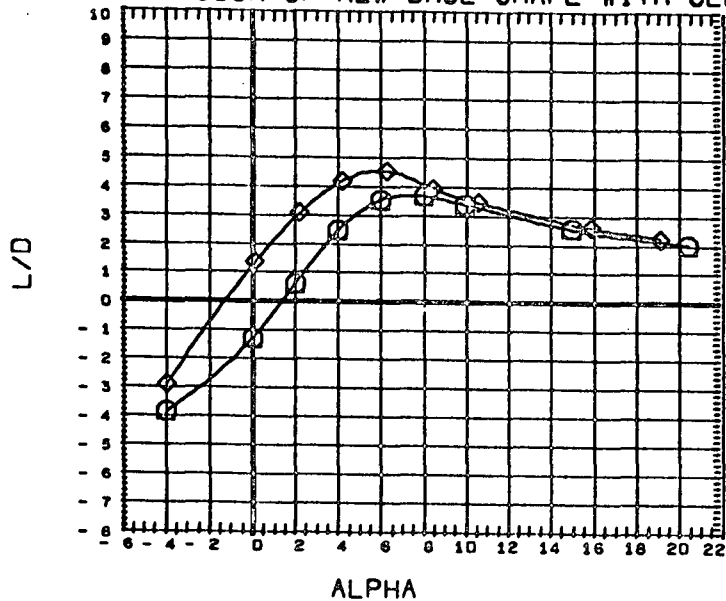
MACH 0.798



BETA ELEVTR CANARD
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 -0.019 0.000 0.000
 0.031 0.000 0.000

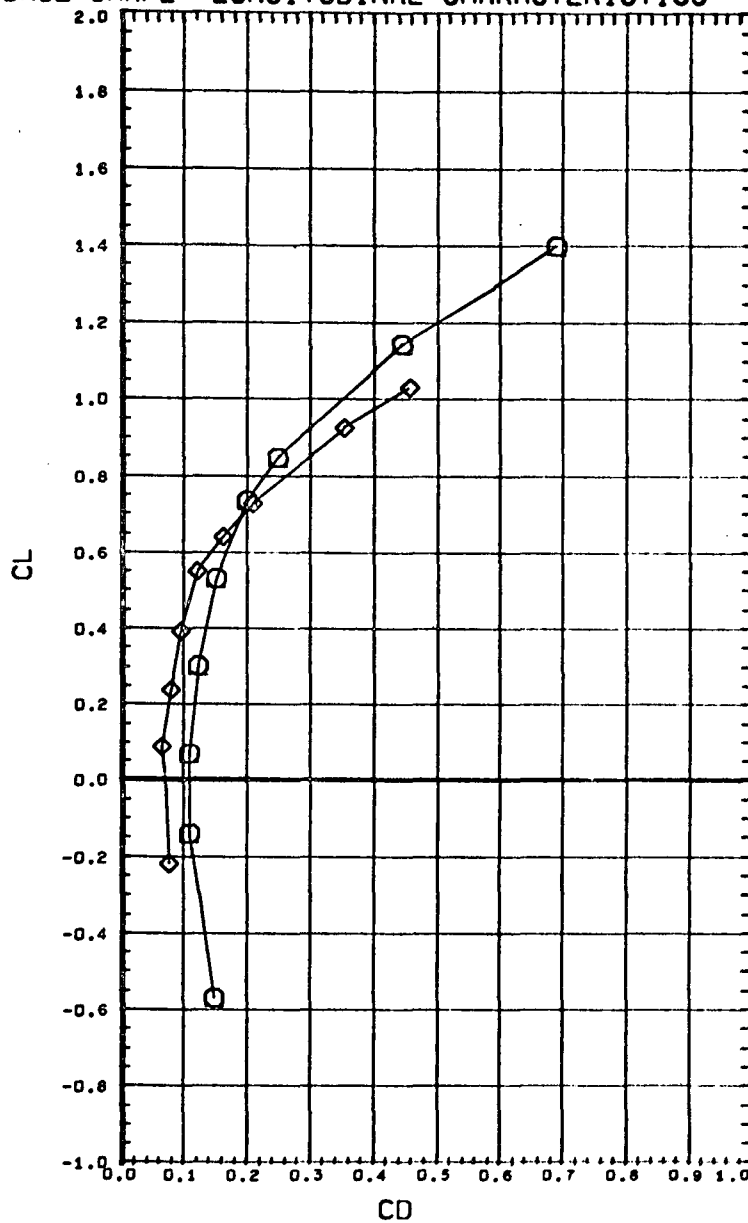
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 BREF 3.4530 FT.
 XMRP 2.5930 FT.
 YMRP 0.0000 FT.
 ZMRP 0.0187 FT.
 SCALE 1.5000 PER CT

COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RU9001) ○ CAL MSFC/LMSC BOOSTER B4C2F2W3V1
 (AN2001) ◇ DATA NOT AVAILABLE FOR ALL CONDITIONS
 (AN3001) ◇ NSRCC-3210, MSFC/LMSC BOOSTER B1C2F2W1V1

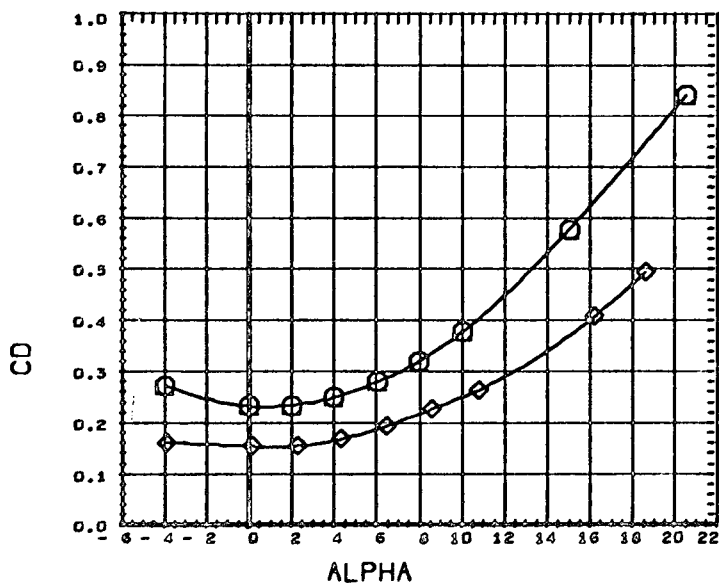
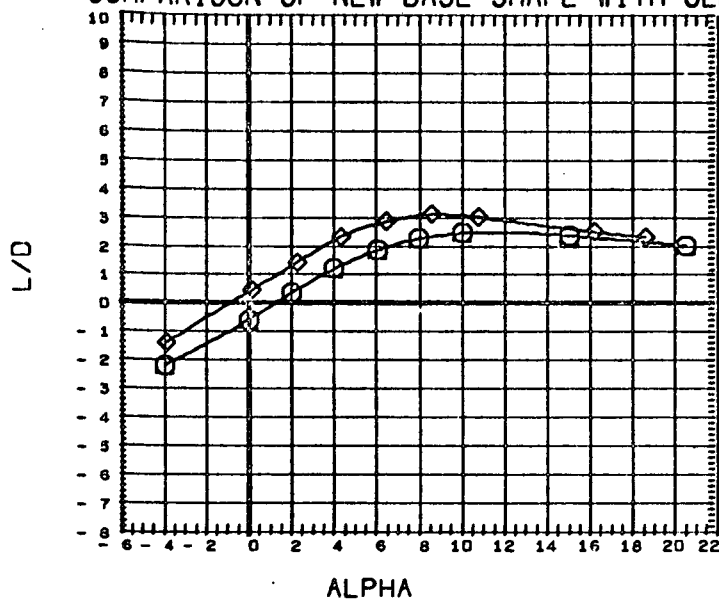
MACH 0.697



BETA ELEVTR CANARD
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 -0.019 0.000 0.000
 0.031 0.000 0.000

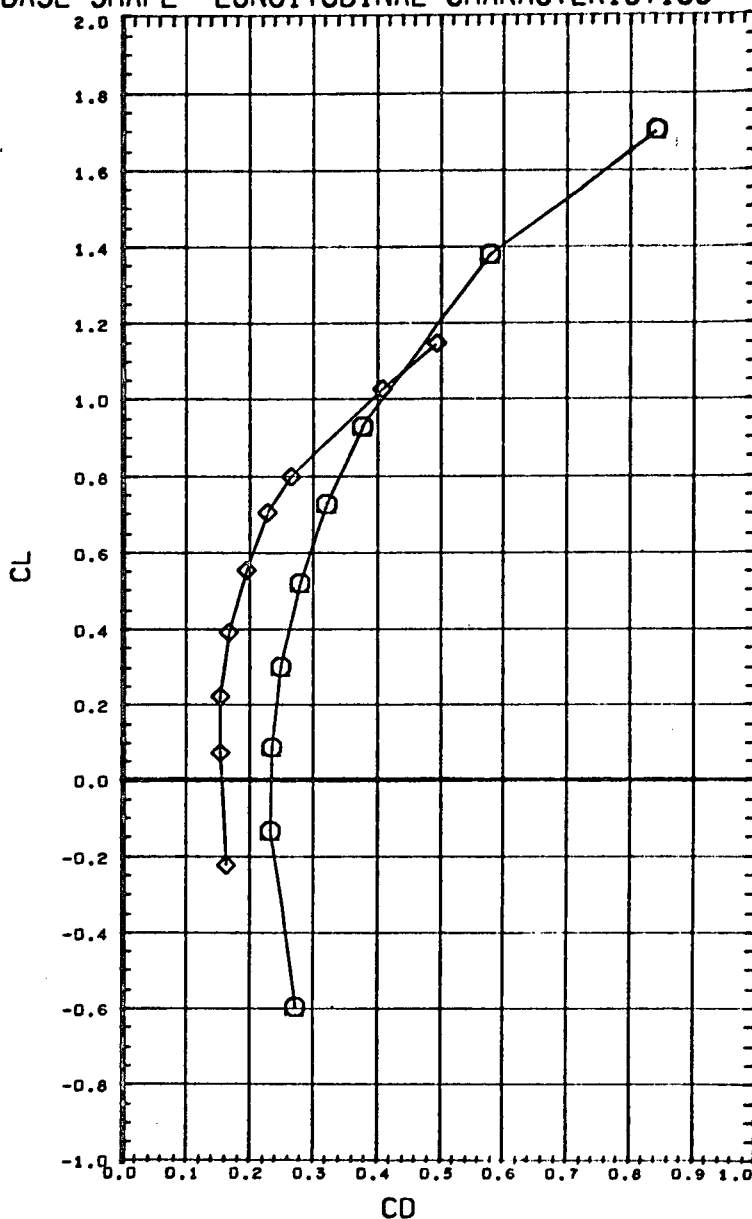
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 LREF 3.4530 FT.
 BREF 3.4530 FT.
 XMRP 2.5930 FT.
 YMRP 0.0000 FT.
 ZMRP 0.0187 FT.
 SCALE 1.5000 PER CT

COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RU9001) ○ CAL MSFC/LMSC BOOSTER B4C2F2W3V1
 (AN2001) △ DATA NOT AVAILABLE FOR ALL CONDITIONS
 (AN3001) ◇ NSRDC-3210,MSFC/LMSC BOOSTER B1C2F2W1V1

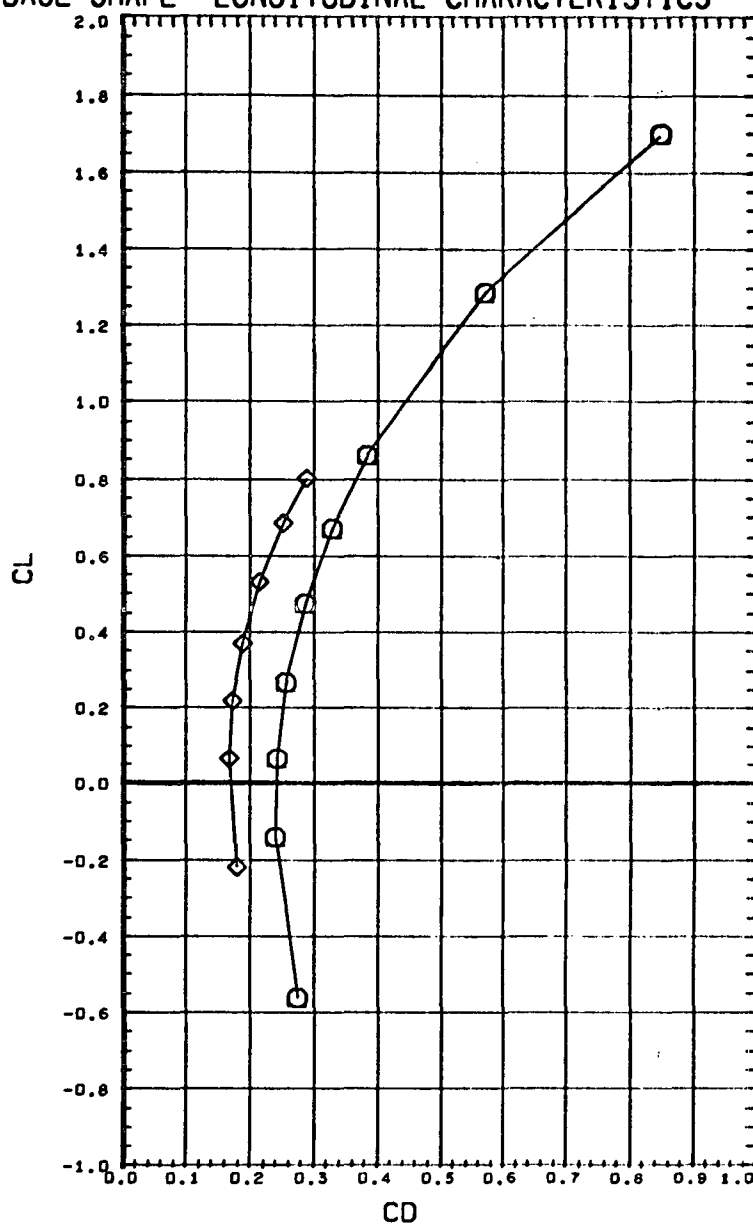
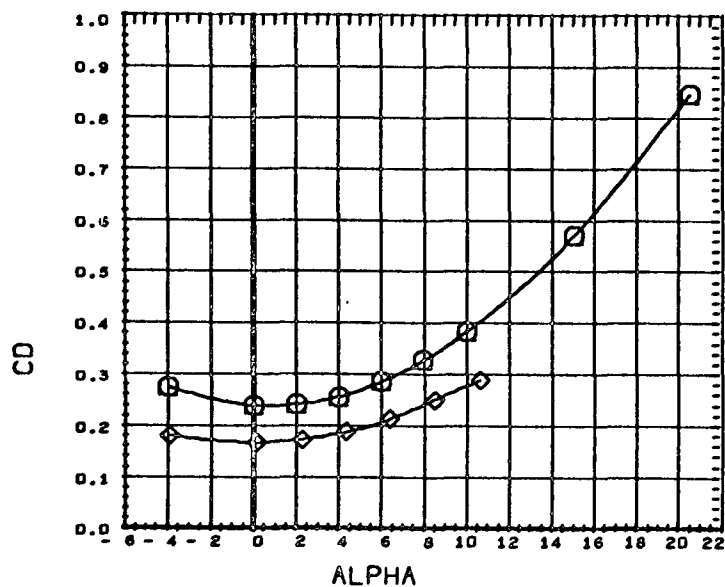
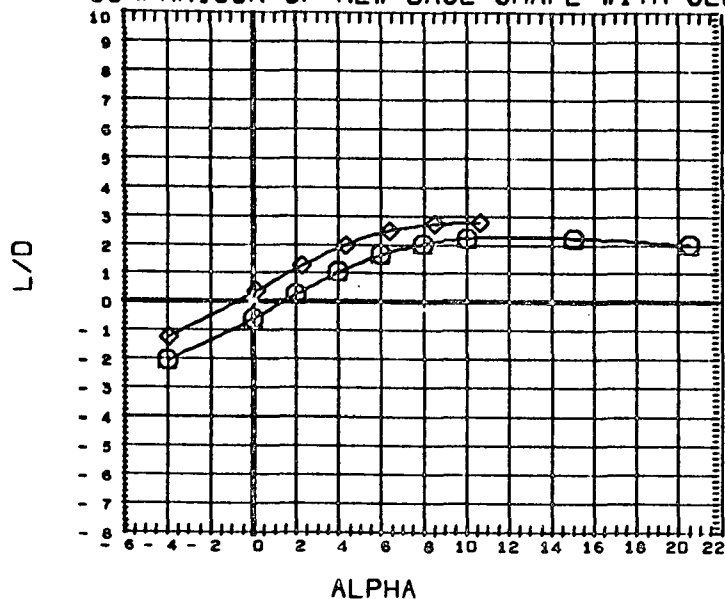
MACH 0.999



BETA ELEVTR CANARD
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 -0.019 0.000 0.000
 0.031 0.000 0.000

REFERENCE INFORMATION
 SREF 1.3550 SQ.FT.
 LREF 3.4530 FT.
 BREF 3.4530 FT.
 XMRP 2.5950 FT.
 YMRP 0.0000 FT.
 ZMRP 0.0187 FT.
 SCALE 1.5000 PER C1

COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS



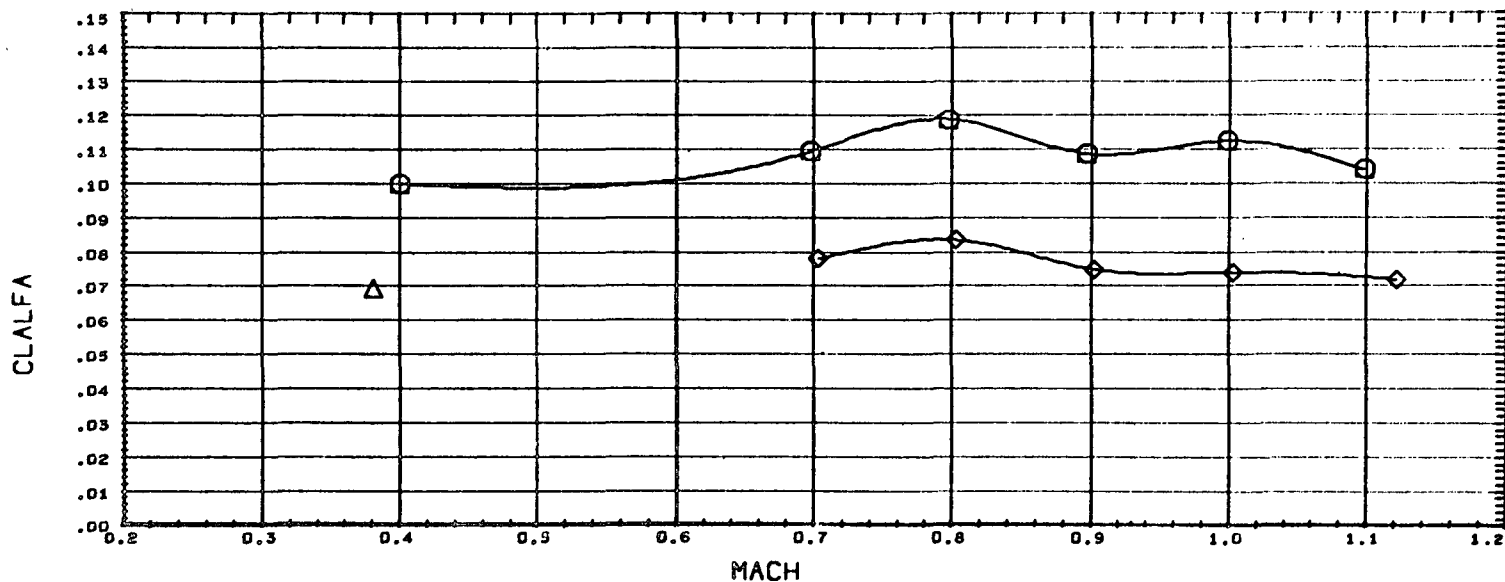
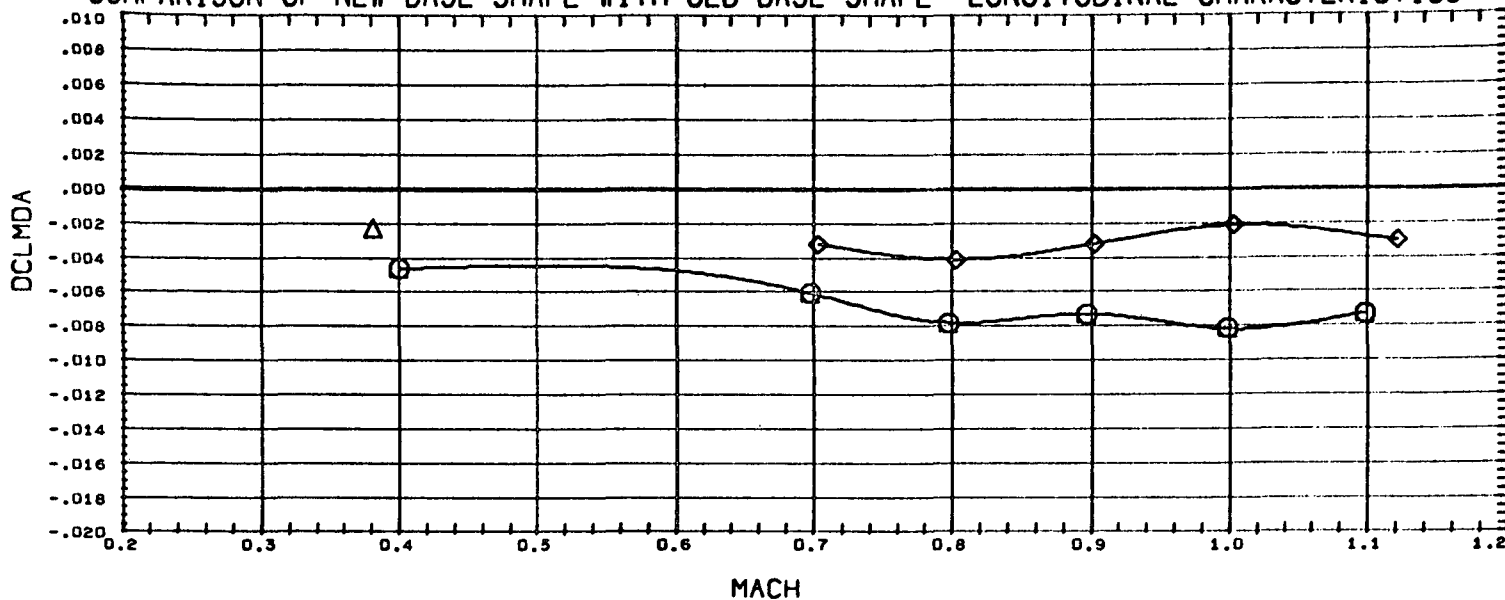
DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RU9001) ○ CAL MSFC/LMSC BOOSTER B4C2F2W3V1
 (AN2001) △ DATA NOT AVAILABLE FOR ALL CONDITIONS
 (AN3001) ◇ NSRCC-3210, MSFC/LMSC BOOSTER B1C2F2W1V1

MACH 1.099

BETA ELEVTR CANARD
 0.000 0.000 0.000
 -0.019 0.000 0.000
 0.031 0.000 0.000

REFERENCE INFORMATION
 SREF 1.3550 SQ.FT.
 LREF 3.4530 FT.
 BREF 3.4530 FT.
 XMRP 2.5950 FT.
 YMRP 0.0000 FT.
 ZMRP 0.0187 FT.
 SCALE 1.5000 PER CT

COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS

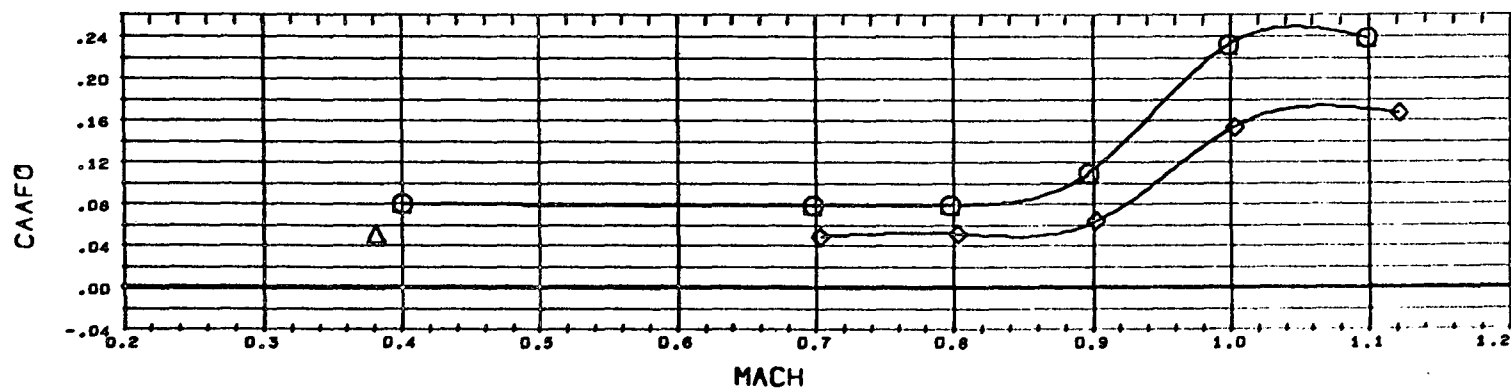
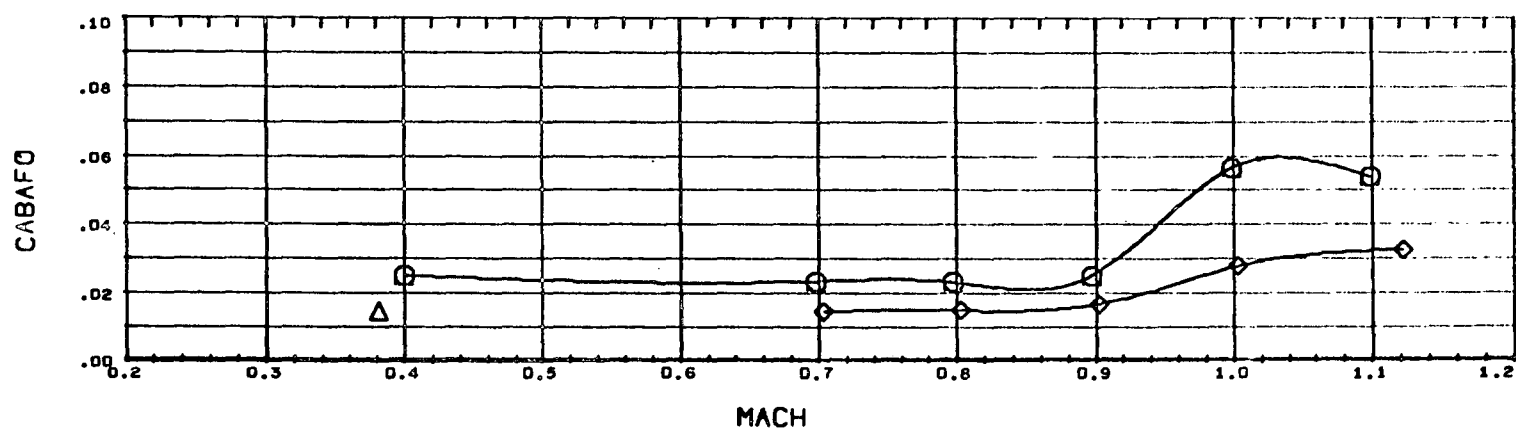
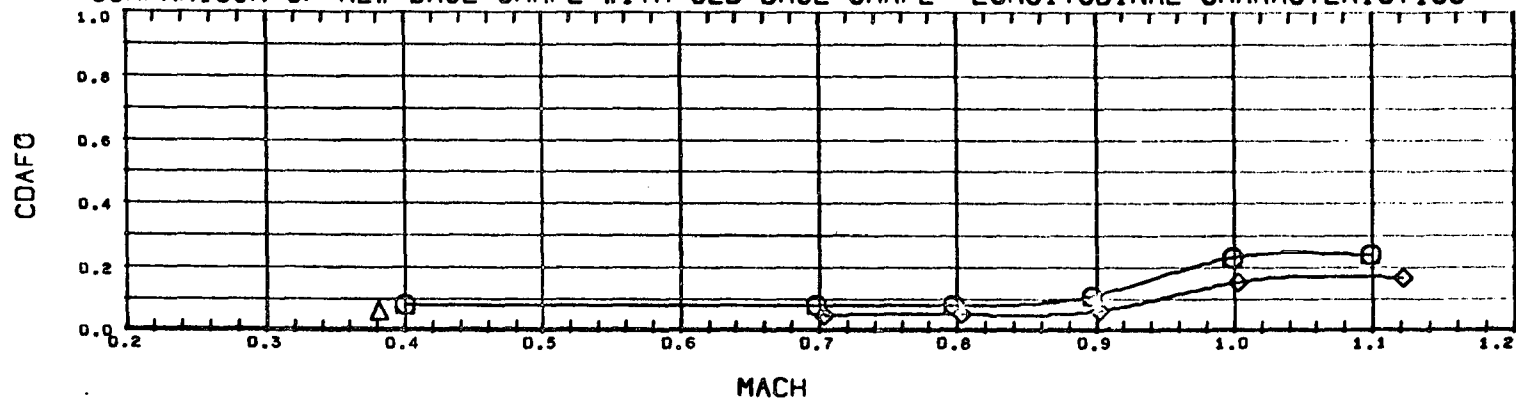


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(R09001)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(AN2001)	NSRDC-3110,MSFC/LMSC BOOSTER B1C2F1W1V1
(AN3001)	NSRDC-3210,MSFC/LMSC BOOSTER B1C2F2W1V1

BETA	ELEVTR	CANARD
0.000	0.000	0.000
-0.019	0.000	0.000
0.031	0.000	0.000

REFERENCE INFORMATION	
SREF	1.3550 SQ.FT.
LREF	3.4530 FT.
BREF	3.4530 FT.
XMRP	2.5950 FT.
YMRP	0.0000 FT.
ZMRP	0.0187 FT.
SCALE	1.5000 PER CT

COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LONGITUDINAL CHARACTERISTICS

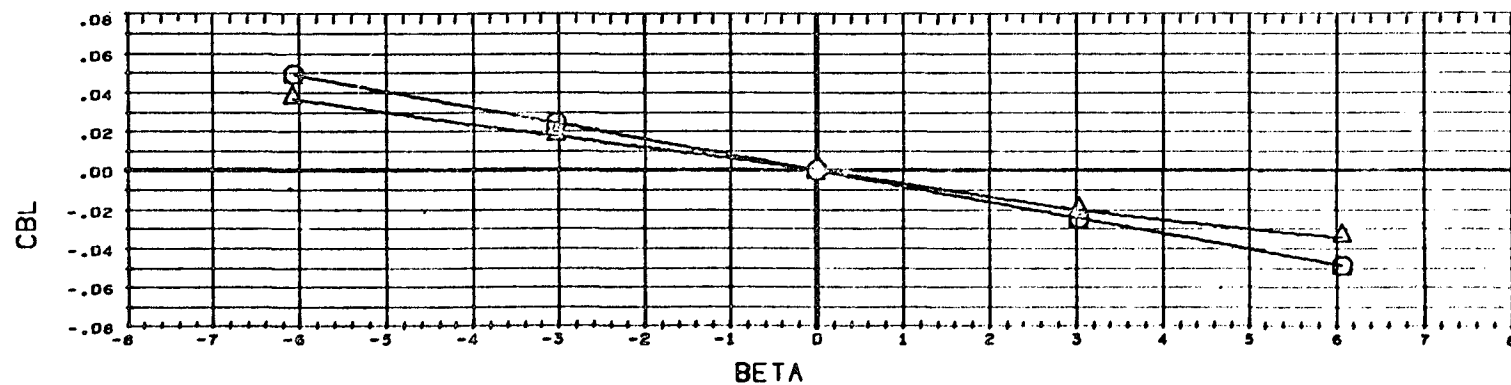
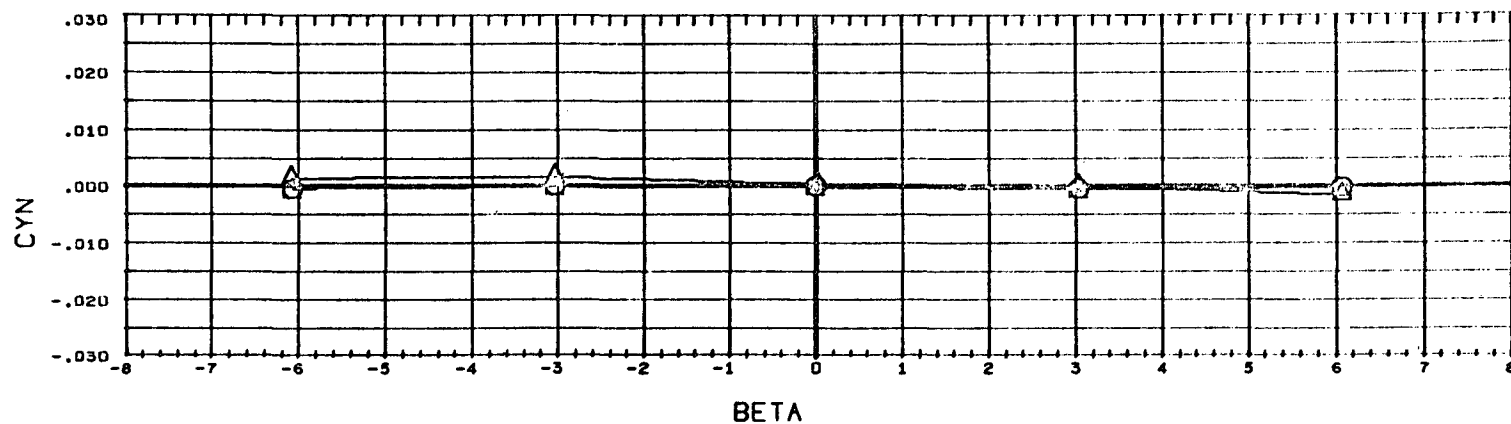
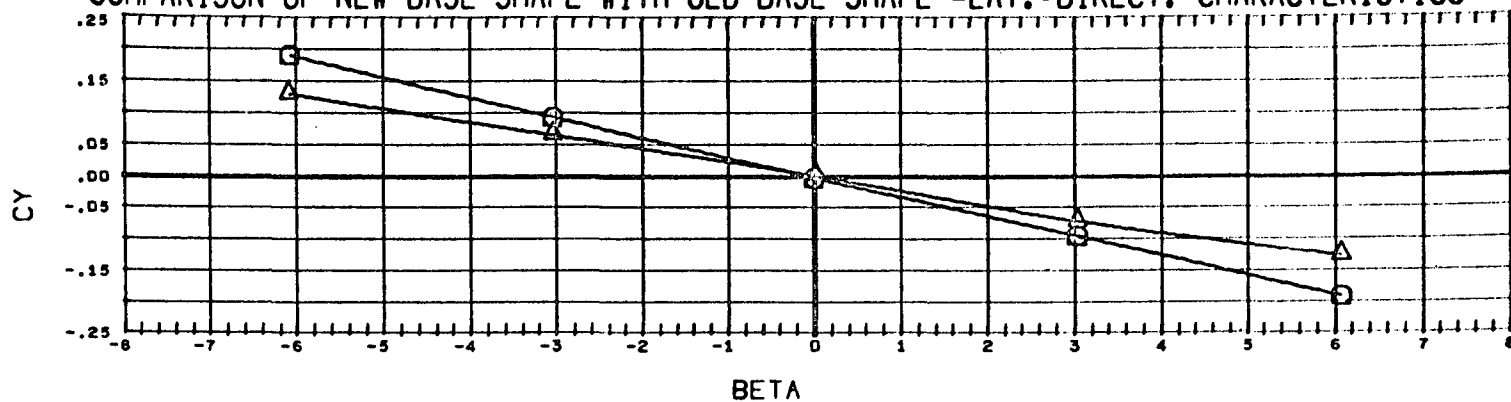


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(B09001)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(B2001)	NSRDC-3110,MSFC/LMSC BOOSTER B1C2F1W1V1
(B3001)	NSRDC-3210,MSFC/LMSC BOOSTER B1C2F2W1V1

BETA	ELEVTR	CANARD
0.000	0.000	0.000
-0.019	0.000	0.000
0.031	0.000	0.000

REFERENCE INFORMATION		
SREF	1.3550	SQ.FT.
LREF	3.4330	FT.
BREF	3.4530	FT.
XMRP	2.5950	FT.
YMRP	0.0000	FT.
ZMRP	0.0187	FT.
SCALE	1.5000	PER CT

COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LAT.-DIRECT. CHARACTERISTICS



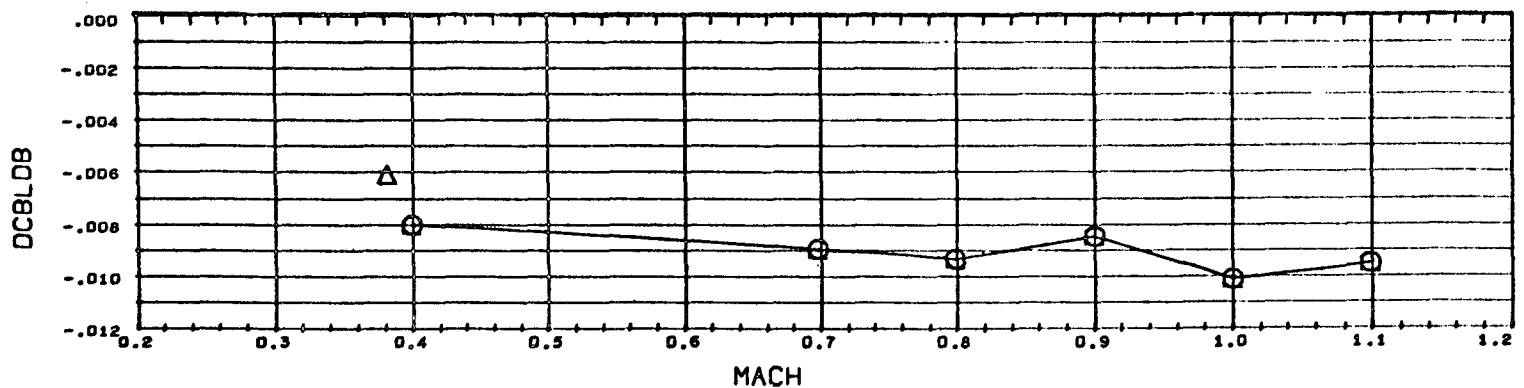
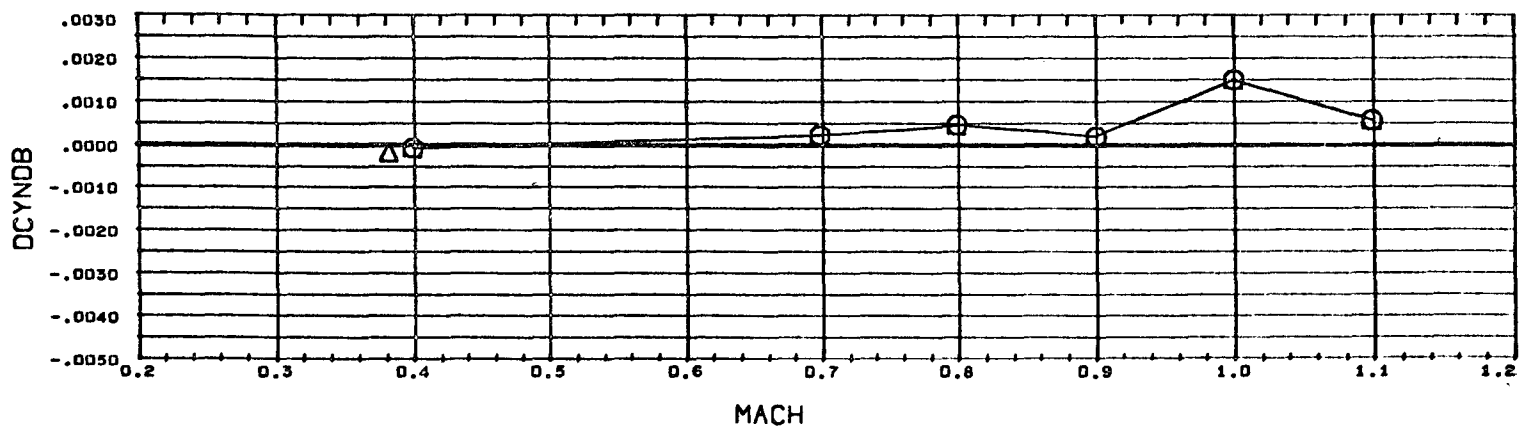
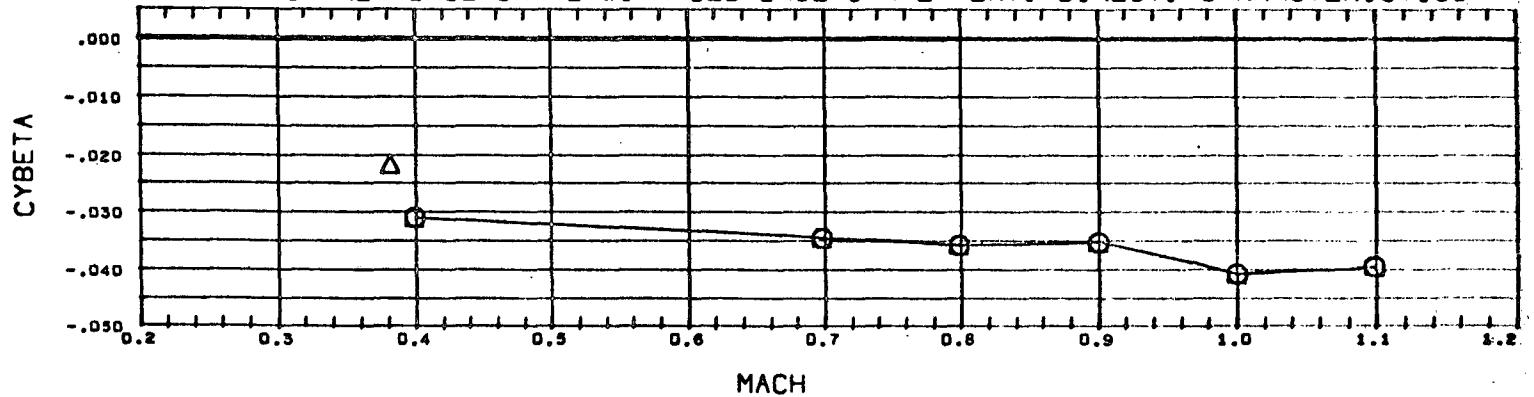
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(CU9002)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(AN2002)	NSRDC-3110,MSFC/LMSC BOOSTER B1C2F1W1V1

ALPHA	ELEVTR	CANARD
0.000	0.000	0.000
0.033	0.000	0.000

REFERENCE INFORMATION	
SREF	1.3550 SQ.FT.
LREF	3.4530 FT.
BREF	3.4530 FT.
XMRP	2.5950 FT.
YMRP	0.0000 FT.
ZMRP	0.0187 FT.
SCALE	1.5000 PER CT

MACH 0.399

COMPARISON OF NEW BASE SHAPE WITH OLD BASE SHAPE -LAT.-DIRECT. CHARACTERISTICS

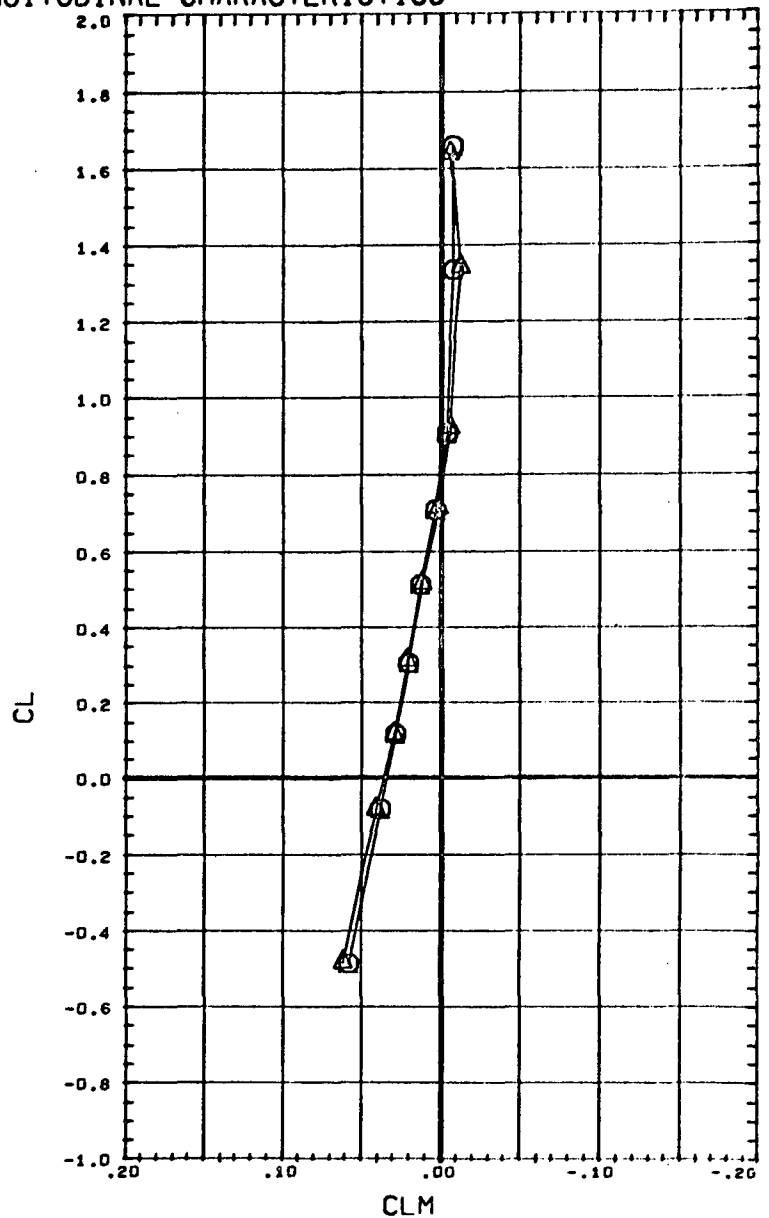
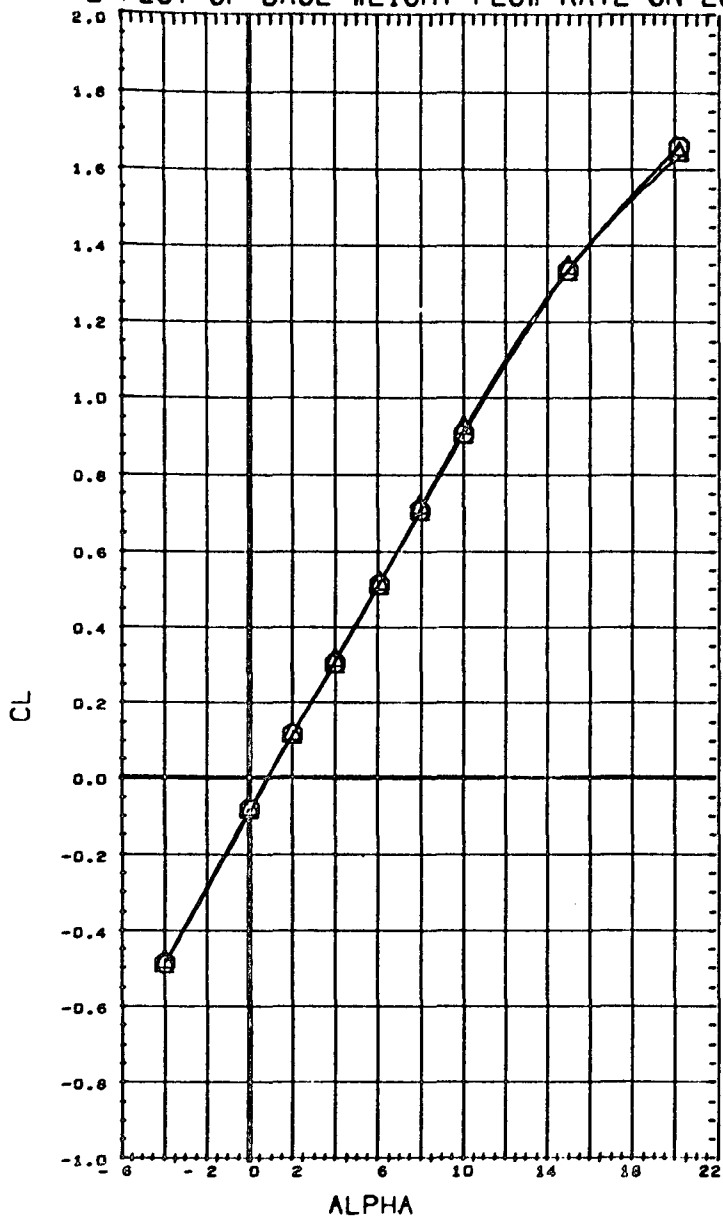


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RU9002)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(AN2002)	NSRDC-3110, MSFC/LMSC BOOSTER B1C2F1W1V1

ALPHA	ELEVTR	CANARD
0.000	0.000	0.000
0.033	0.000	0.000

REFERENCE INFORMATION		
SREF	1.3550	80.FT.
LREF	3.4530	FT.
BREF	3.4530	FT.
XMRP	2.5950	FT.
YMRP	0.0000	FT.
ZMRP	0.0187	FT.
SCALE	1.5000	PER CT

EFFECT OF BASE WEIGHT FLOW RATE ON LONGITUDINAL CHARACTERISTICS

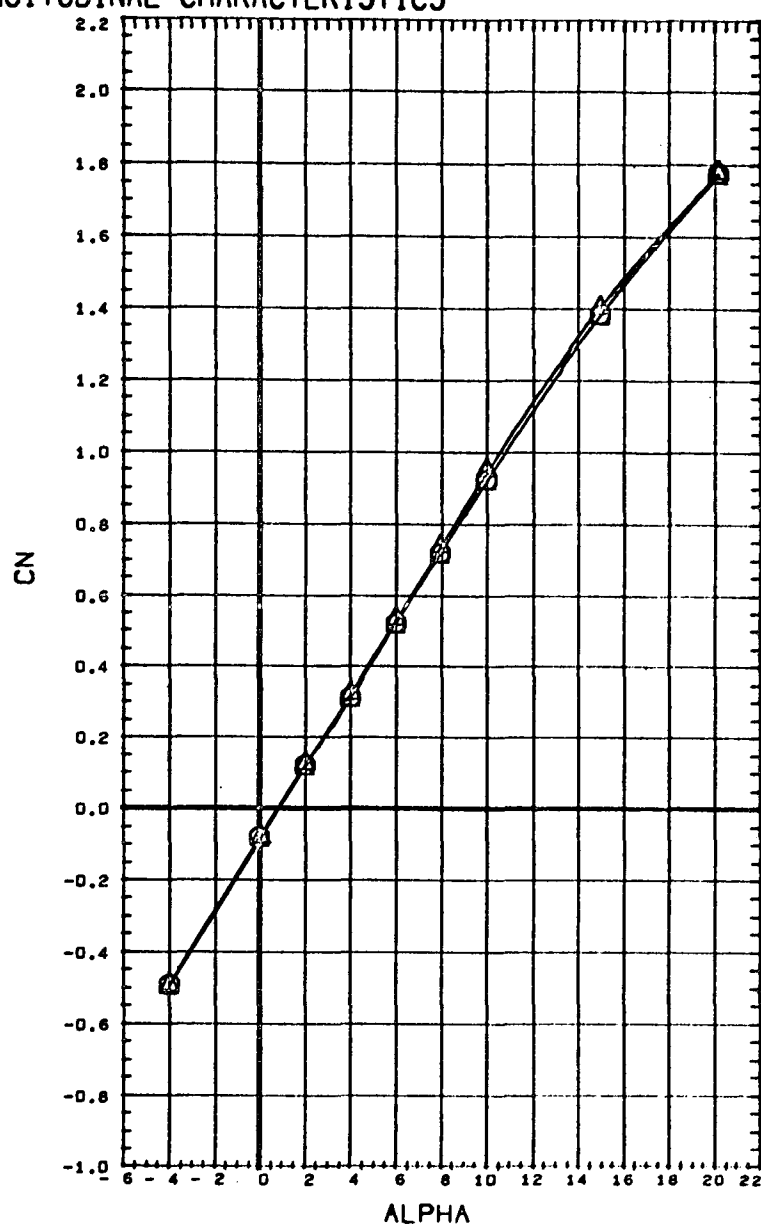
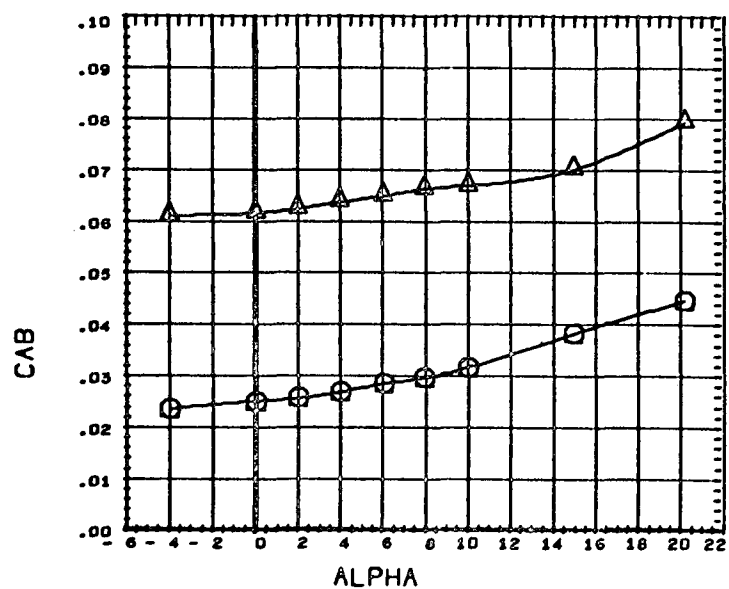
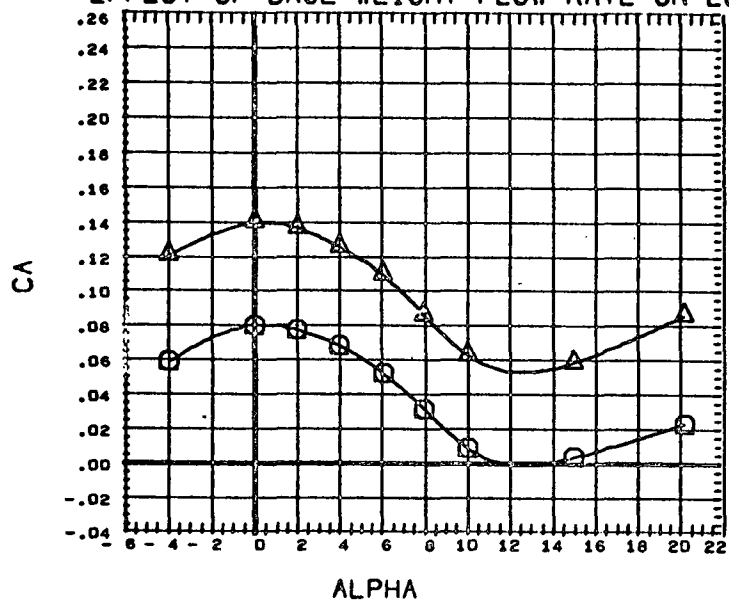


DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CU9101) CAL MSFC/LMSC BOOSTER B4C2F2W3V1
 (CU9057) CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH 0.400

BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	0.000	0.353	LREF 3.4530 FT.
				BREF 3.4530 FT.
				XMRF 2.5950 FT.
				YMRF 0.0000 FT.
				ZMRF 0.0187 FT.
				SCALE 1.5000 PER CT

EFFECT OF BASE WEIGHT FLOW RATE ON LONGITUDINAL CHARACTERISTICS

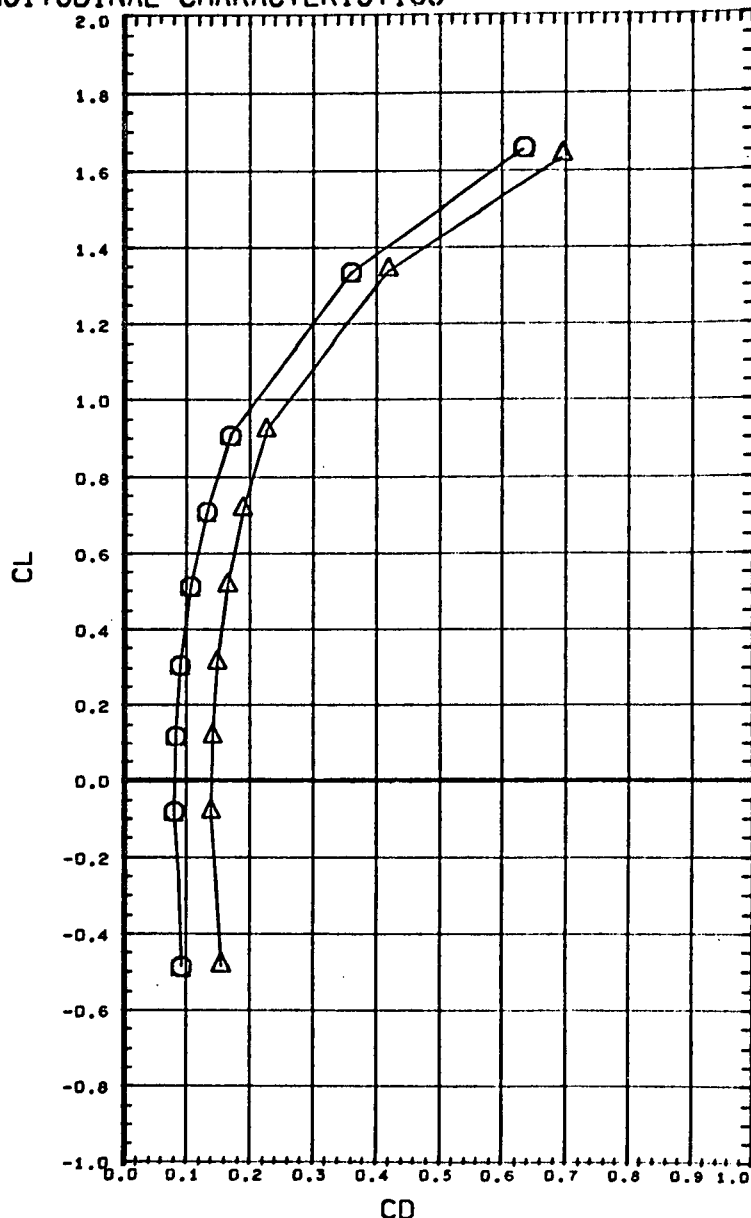
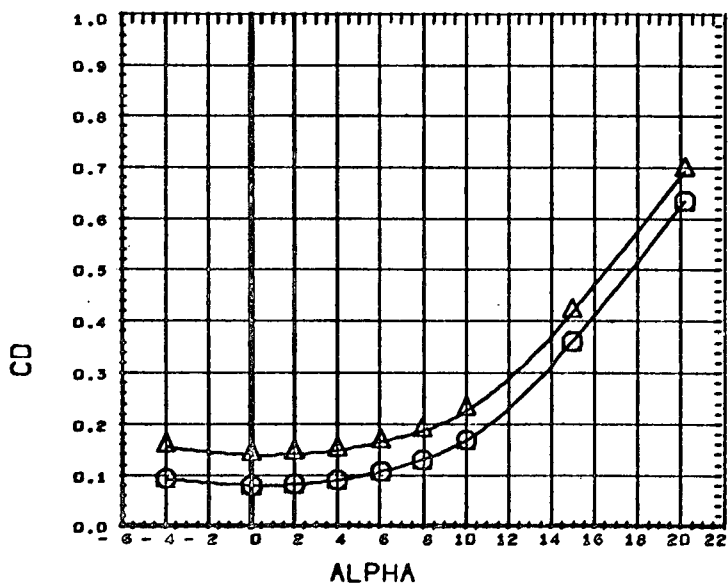
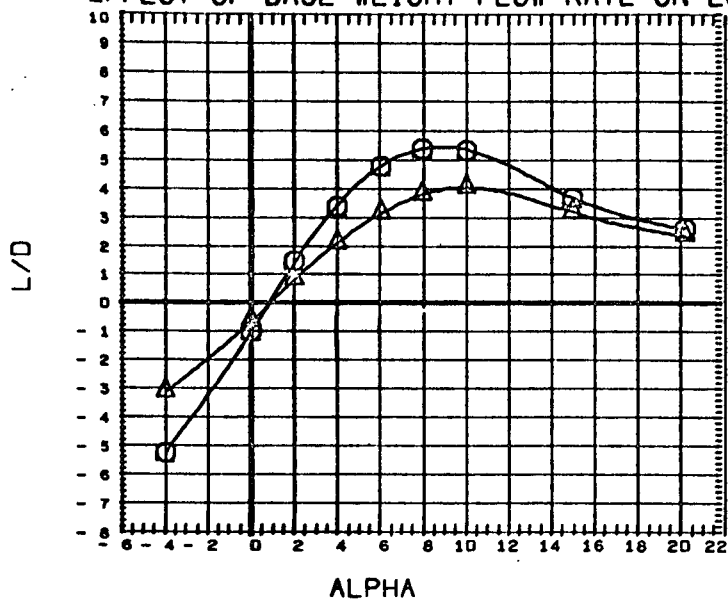


DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CU9001) CAL MSFC/LMSC BOOSTER B4C2F2W3V1
 (CU9057) CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH 0.400

BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	0.000	0.353	LREF 3.4530 FT.
				BREF 3.4530 FT.
				XMRP 2.5950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0107 FT.
				SCALE 1.5000 PER CT

EFFECT OF BASE WEIGHT FLOW RATE ON LONGITUDINAL CHARACTERISTICS

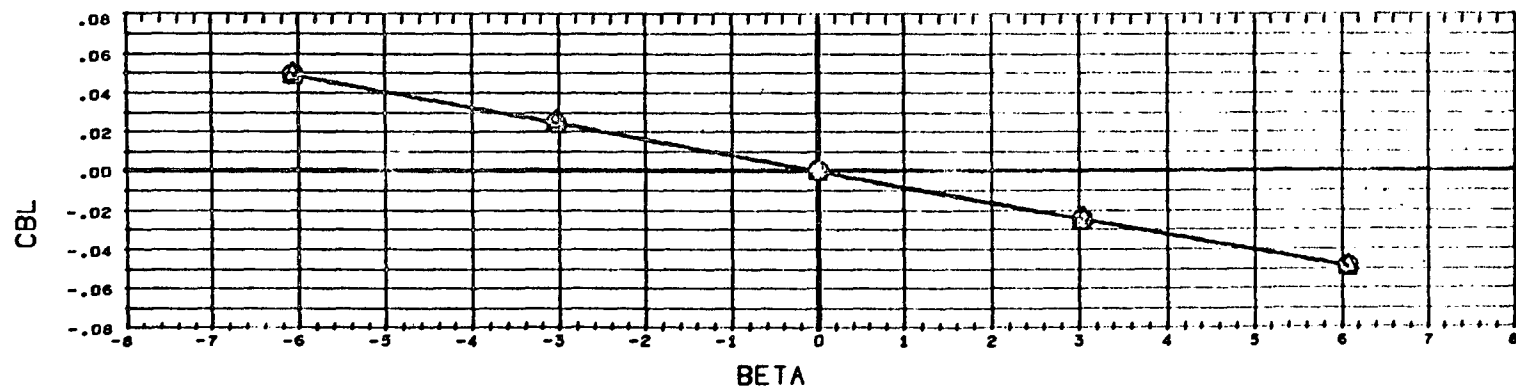
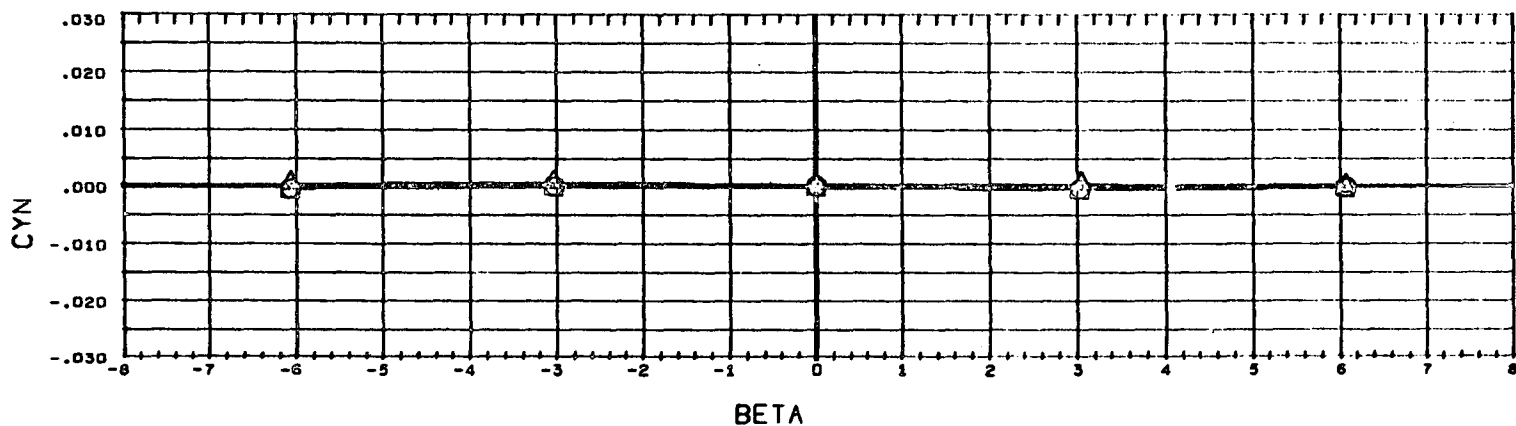
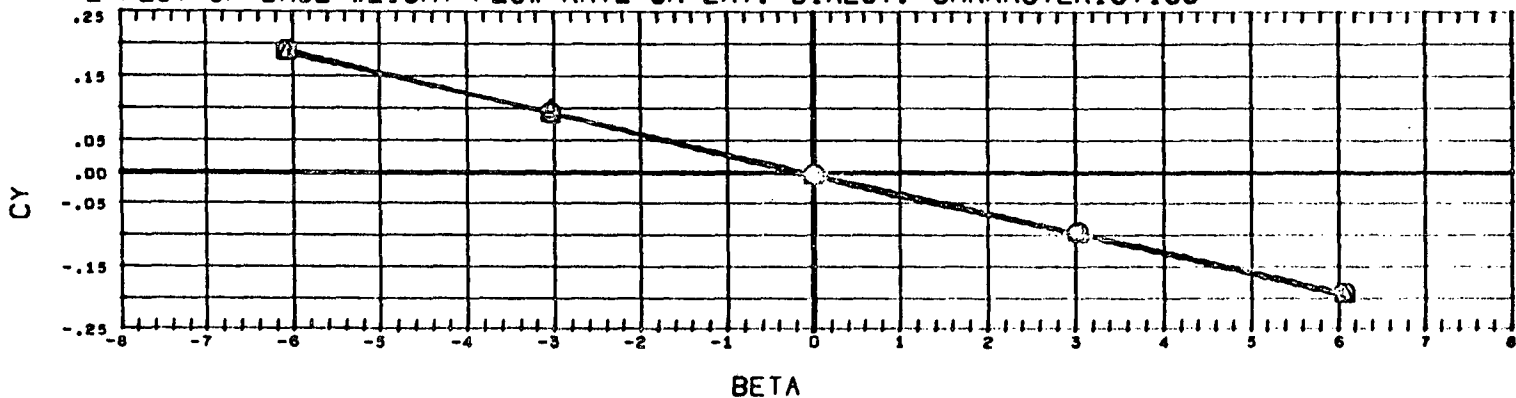


DATA SET SYMBOL CONFIGURATION DESCRIPTION
(CU9001) CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(CU9057) CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH 0.400

BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	0.000	0.353	LREF 3.4530 FT.
				BREF 3.4530 FT.
				XMRP 2.5950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0187 FT.
				SCALE 1.5000 PER CT

EFFECT OF BASE WEIGHT FLOW RATE ON LAT.-DIRECT. CHARACTERISTICS

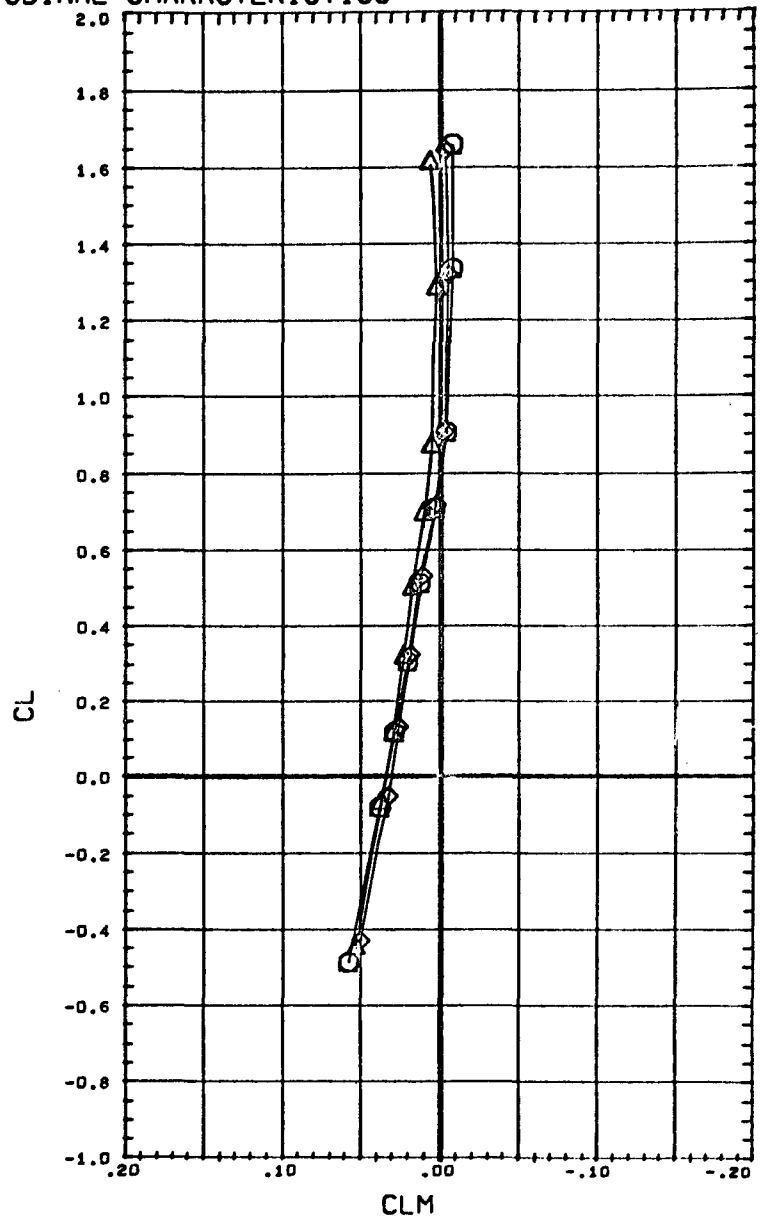
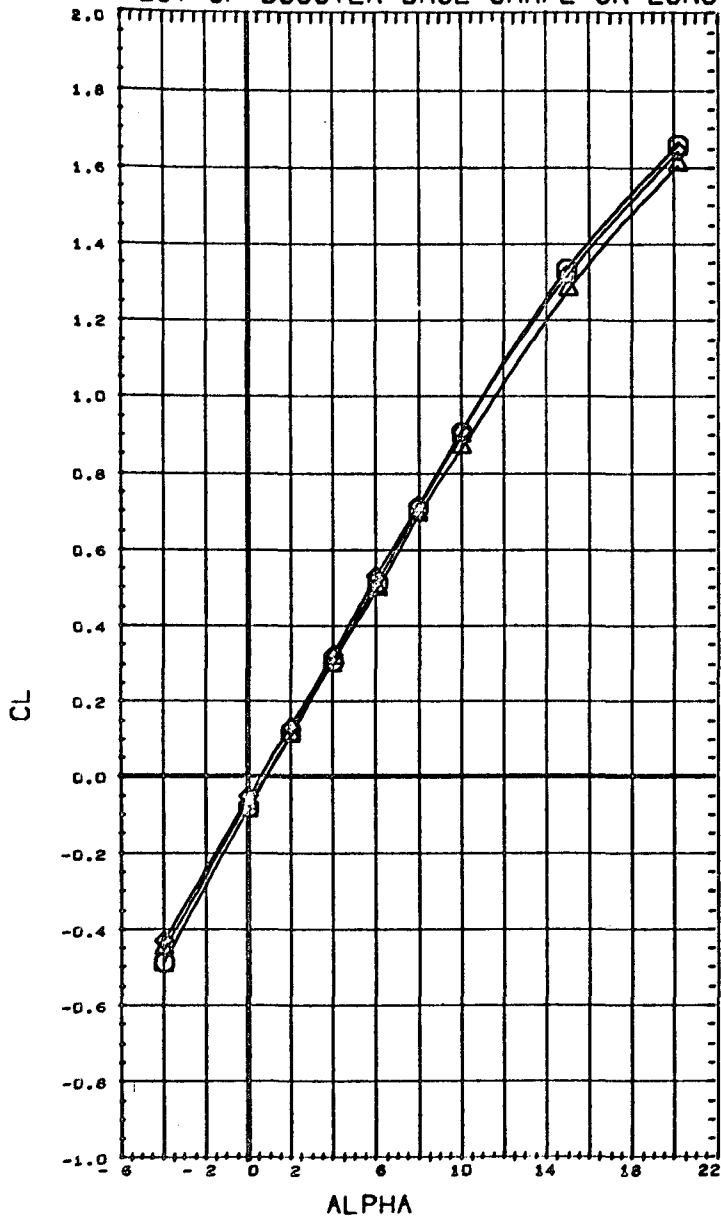


DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CU9002) CAL MSFC/LMSC BOOSTER B4C2F2W3V1
 (RU9058) CAL MSFC/LMSC BOOSTER B4C2F2W3V1

ALPHA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION	
0.000	0.000	0.000	0.000	SREF	1.3550 SQ.FT.
0.000	0.000	0.000	0.353	LREF	3.4530 FT.
				BREF	3.4530 FT.
				XMRP	2.5950 FT.
				YMRP	0.0000 FT.
				ZMRP	0.0187 FT.
				SCALE	1.5000 PER CT

MACH 0.399

EFFECT OF BOOSTER BASE SHAPE ON LONGITUDINAL CHARACTERISTICS

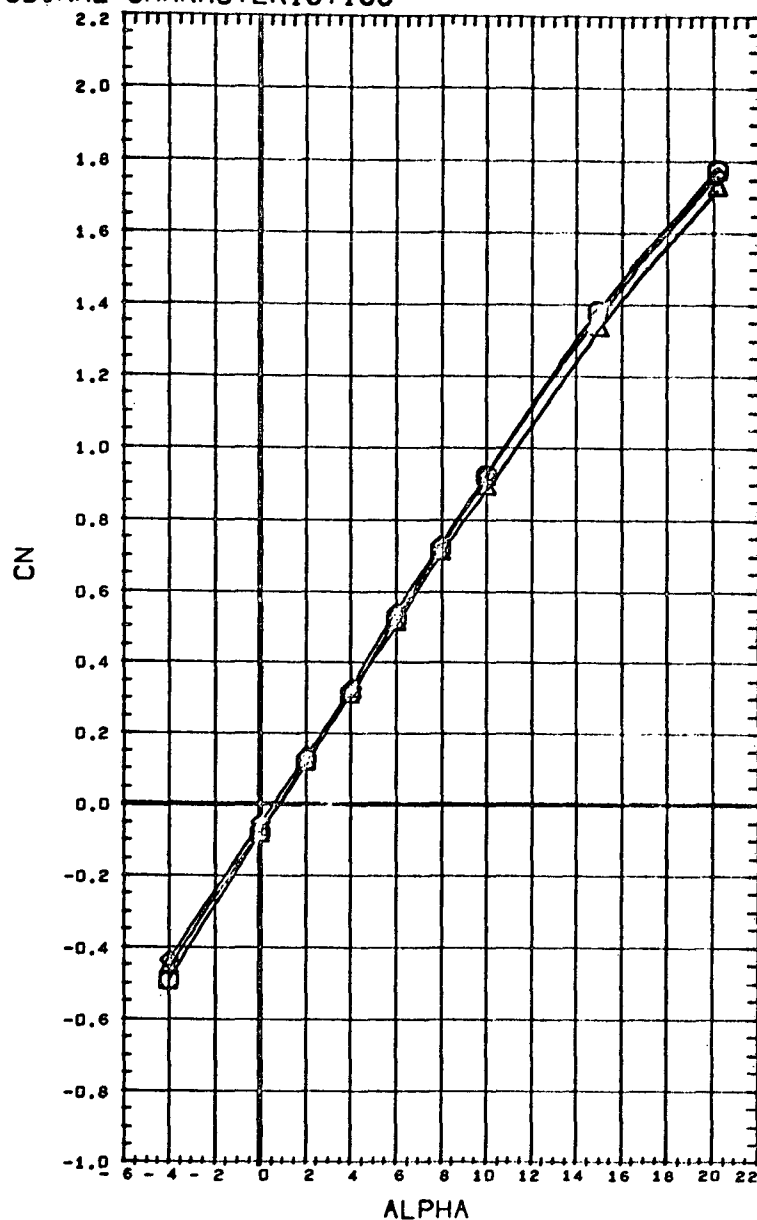
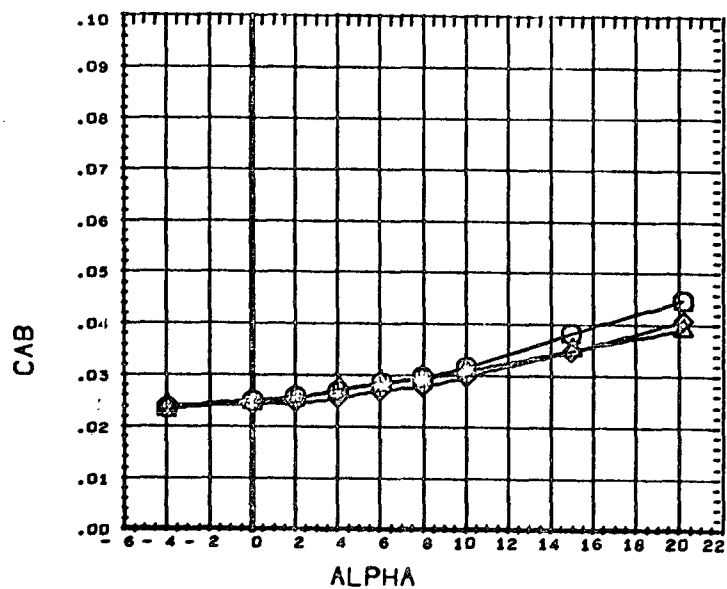
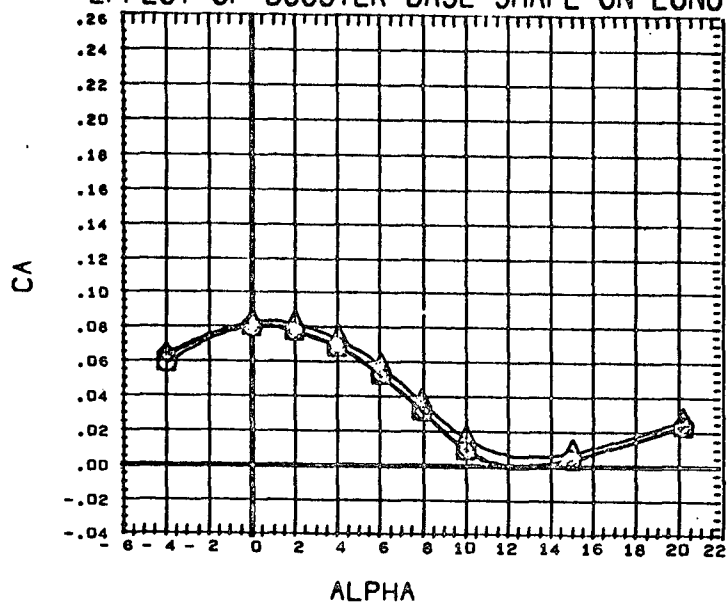


DATA SET	SYMBOL	CONFIGURATION DESCRIPTION
(CU9001)	○	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9005)	△	CAL MSFC/LMSC BOOSTER B5C2F2W3V1 BASE FLAP
(RU9009)	◇	CAL MSFC/LMSC BOOSTER B6C2F2W3V1 BASE VENT

MACH 0.400

BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	0.000	0.000	LREF 3.4530 FT.
0.000	0.000	0.000	0.000	BREF 3.4530 FT.
				XMRP 2.5950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0187 FT.
				SCALE 1.5000 PER CT

EFFECT OF BOOSTER BASE SHAPE ON LONGITUDINAL CHARACTERISTICS

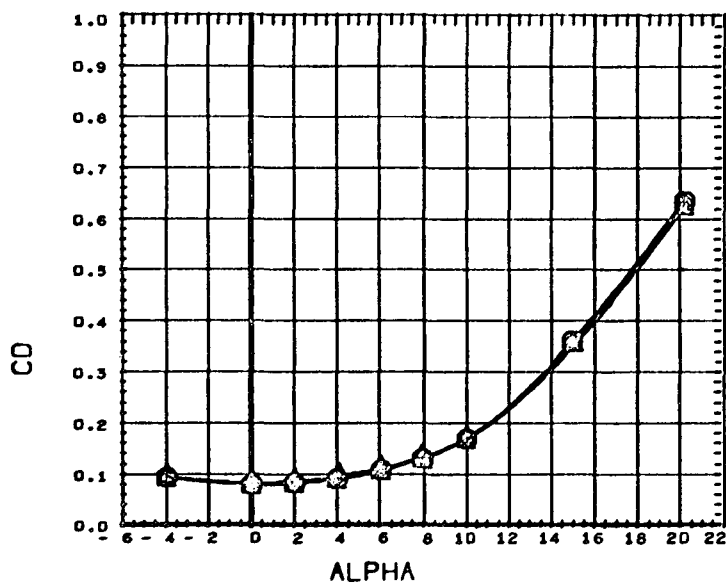
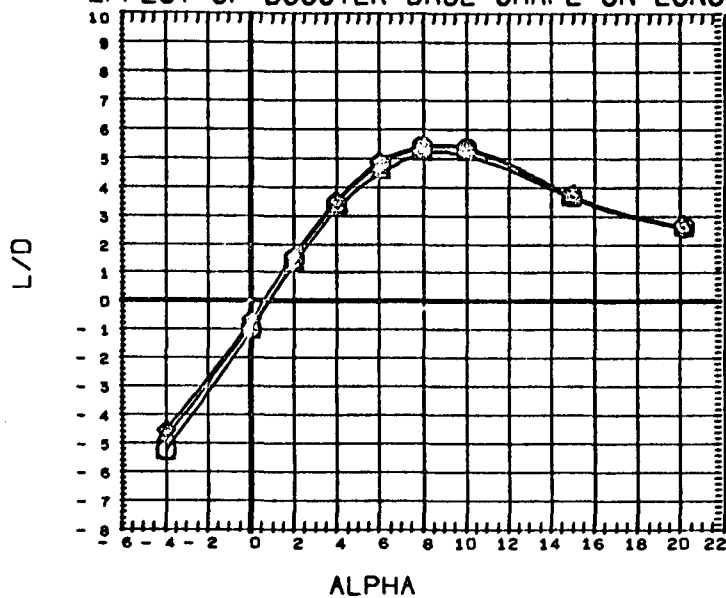


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(CU9001)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9005)	CAL MSFC/LMSC BOOSTER B5C2F2W3V1 BASE FLAP
(RU9009)	CAL MSFC/LMSC BOOSTER B6C2F2W3V1 BASE VENT

MACH 0.400

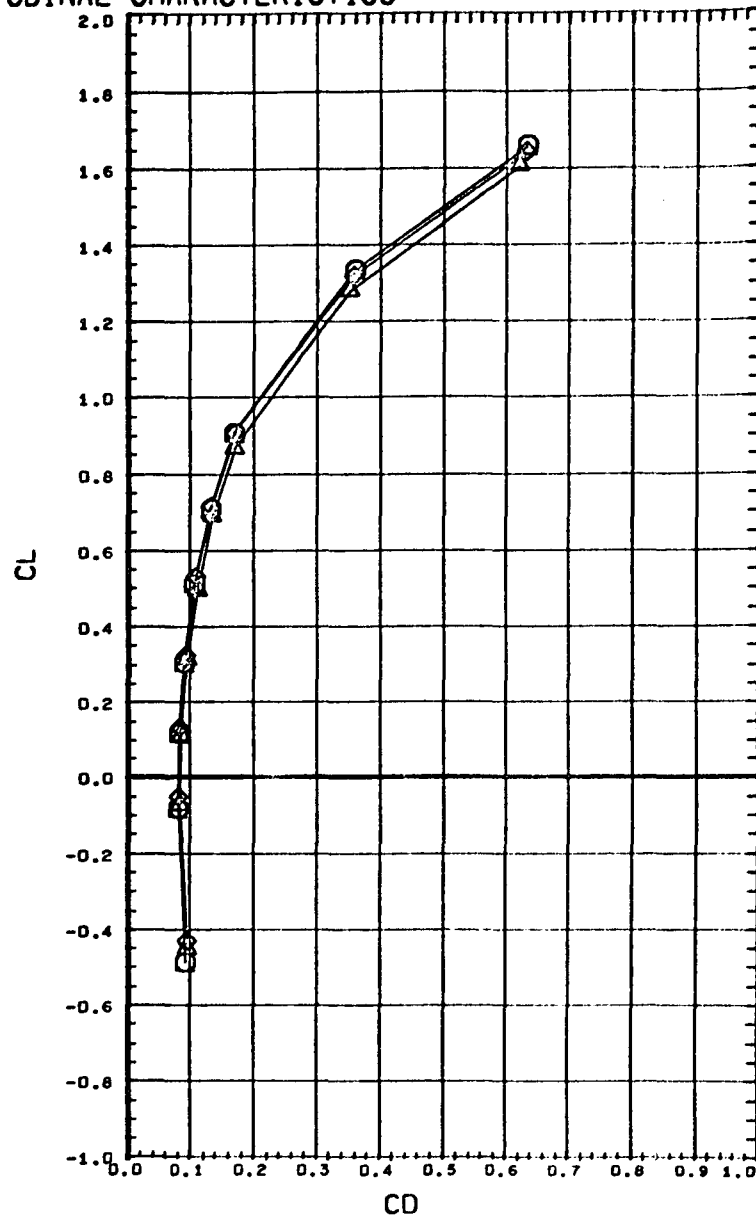
BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	0.000	0.000	LREF 3.4530 FT.
0.000	0.000	0.000	0.000	BREF 3.4530 FT.
				XMRF 2.5950 FT.
				YMRF 0.0000 FT.
				ZMRF 0.0167 FT.
				SCALE 1.5000 PER CT

EFFECT OF BOOSTER BASE SHAPE ON LONGITUDINAL CHARACTERISTICS



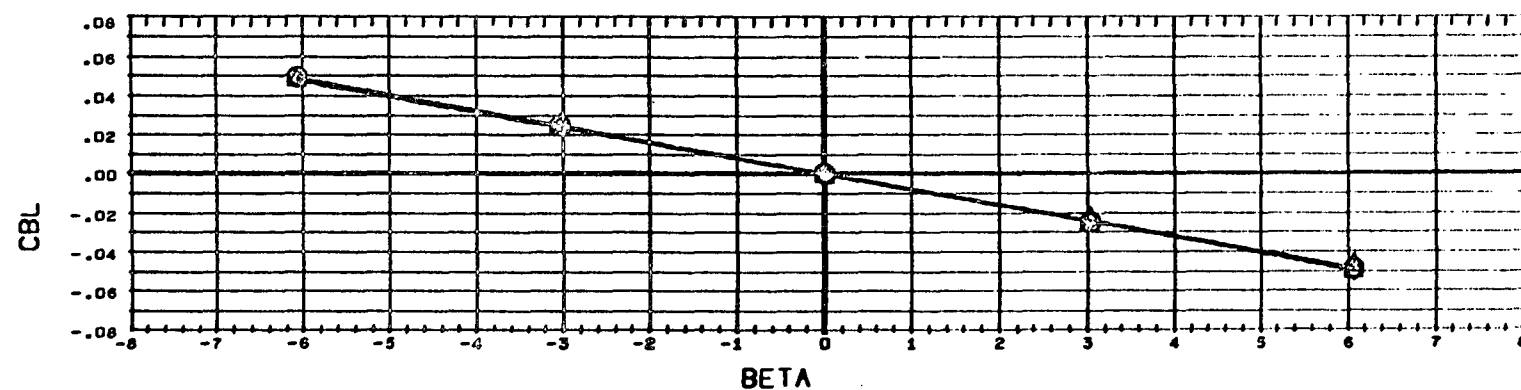
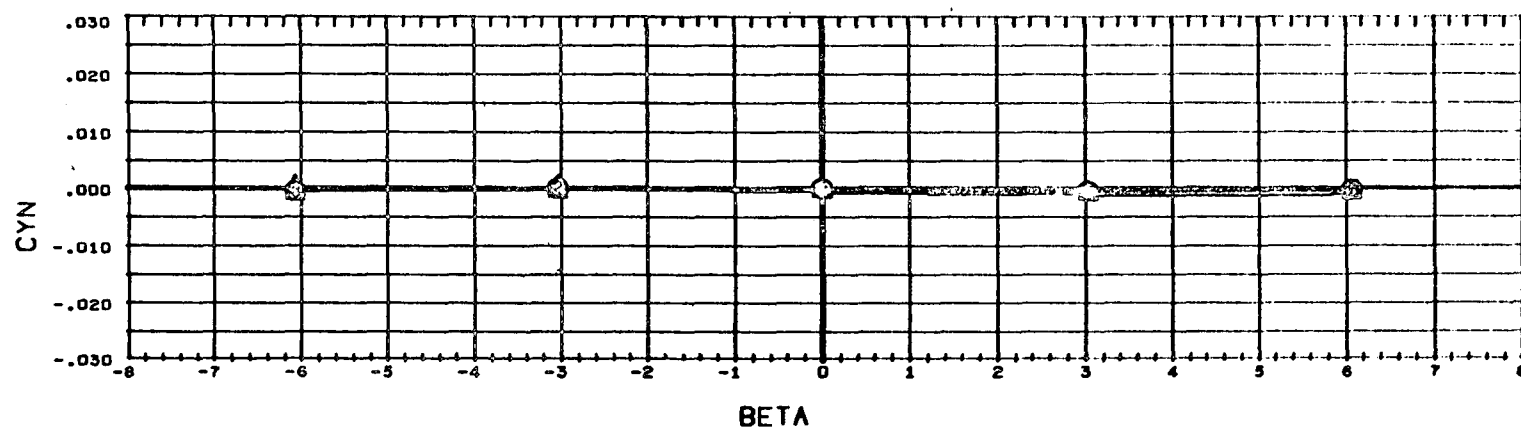
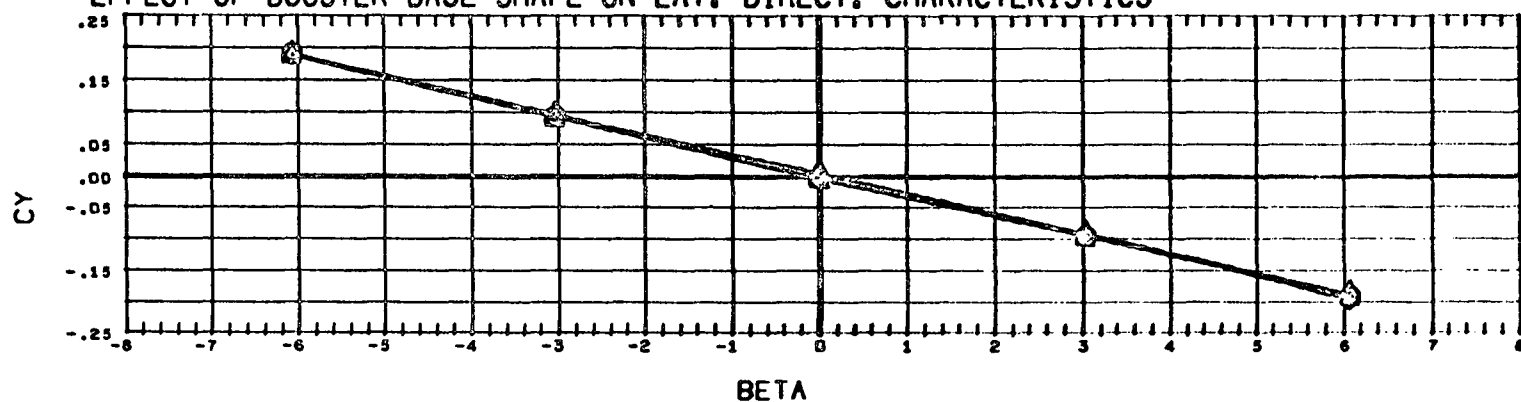
DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(CU9001)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9005)	CAL MSFC/LMSC BOOSTER B5C2F2W3V1 BASE FLAP
(RU9009)	CAL MSFC/LMSC BOOSTER B6C2F2W3V1 BASE VENT

MACH 0.400



BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	0.000	0.000	LREF 3.4530 FT.
0.000	0.000	0.000	0.000	BREF 3.4530 FT.
				XMRF 2.5950 FT.
				YMRF 0.0000 FT.
				ZMRF 0.0187 FT.
				SCALE 1.5000 PER CT

EFFECT OF BOOSTER BASE SHAPE ON LAT.-DIRECT. CHARACTERISTICS

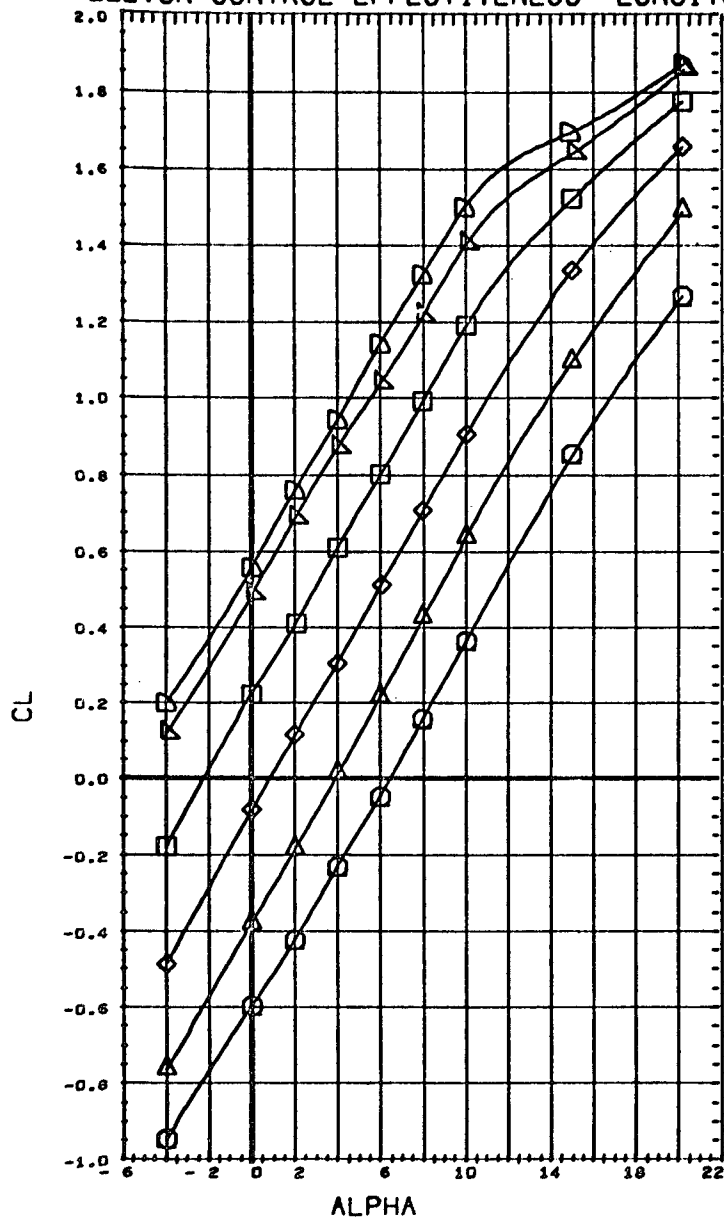


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(CU9002)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9006)	CAL MSFC/LMSC BOOSTER B5C2F2W3V1 BASE FLAP
(RU9010)	CAL MSFC/LMSC BOOSTER B6C2F2W3V1 BASE VENT

ALPHA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 1.3550 30.FT.
0.000	0.000	0.000	0.000	LREF 3.4550 FT.
0.000	0.000	0.000	0.000	BREF 3.4550 FT.
				XMRP 2.5950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0187 FT.
				SCALE 1.5000 PER CT

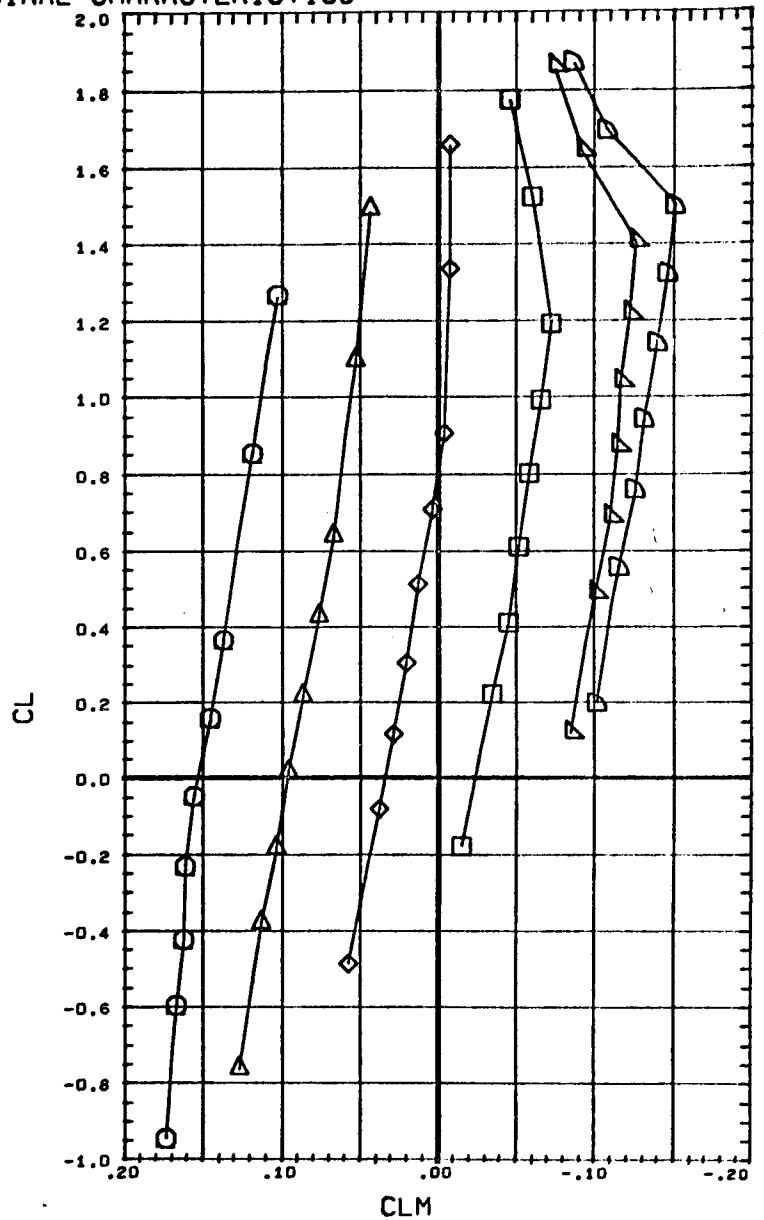
MACH 0.399

ELEVON CONTROL EFFECTIVENESS- LONGITUDINAL CHARACTERISTICS



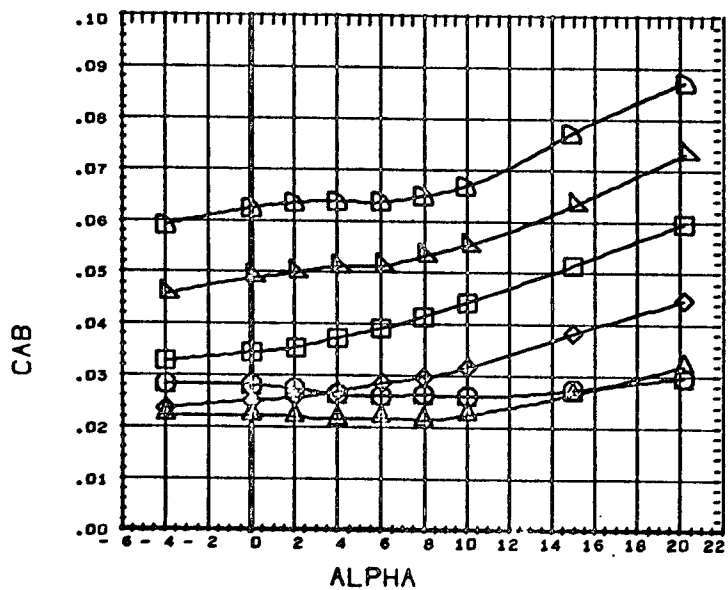
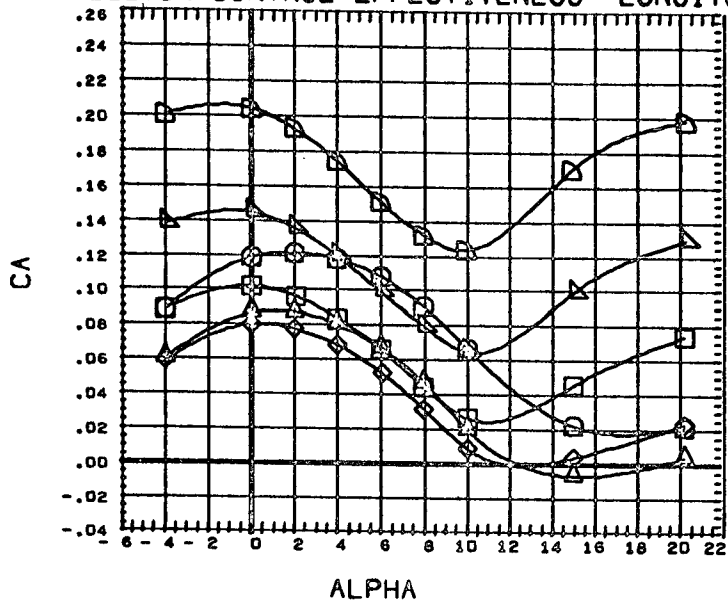
DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RU9053)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(TU9017)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(CU9001)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9013)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9021)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9025)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH 0.398



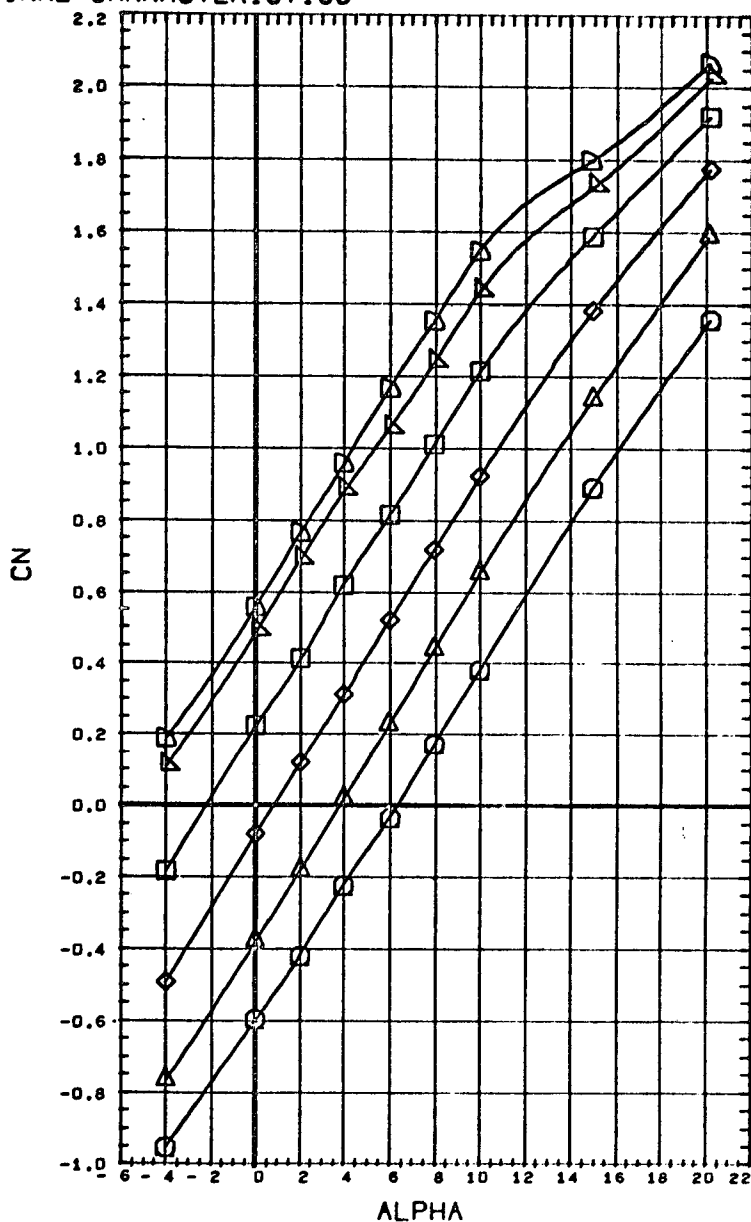
BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	-20.000	0.000	0.000	SREF 1.3550 SQ.FT.
0.000	-10.000	0.000	0.000	LREF 3.4530 FT.
0.000	0.000	0.000	0.000	BREF 3.4530 FT.
0.000	10.000	0.000	0.000	XMRP 2.5950 FT.
0.000	20.000	0.000	0.000	YMRP 0.0000 FT.
0.000	30.000	0.000	0.000	ZMRP 0.0187 FT.
				SCALE 1.5000 PER CT

ELEVON CONTROL EFFECTIVENESS- LONGITUDINAL CHARACTERISTICS



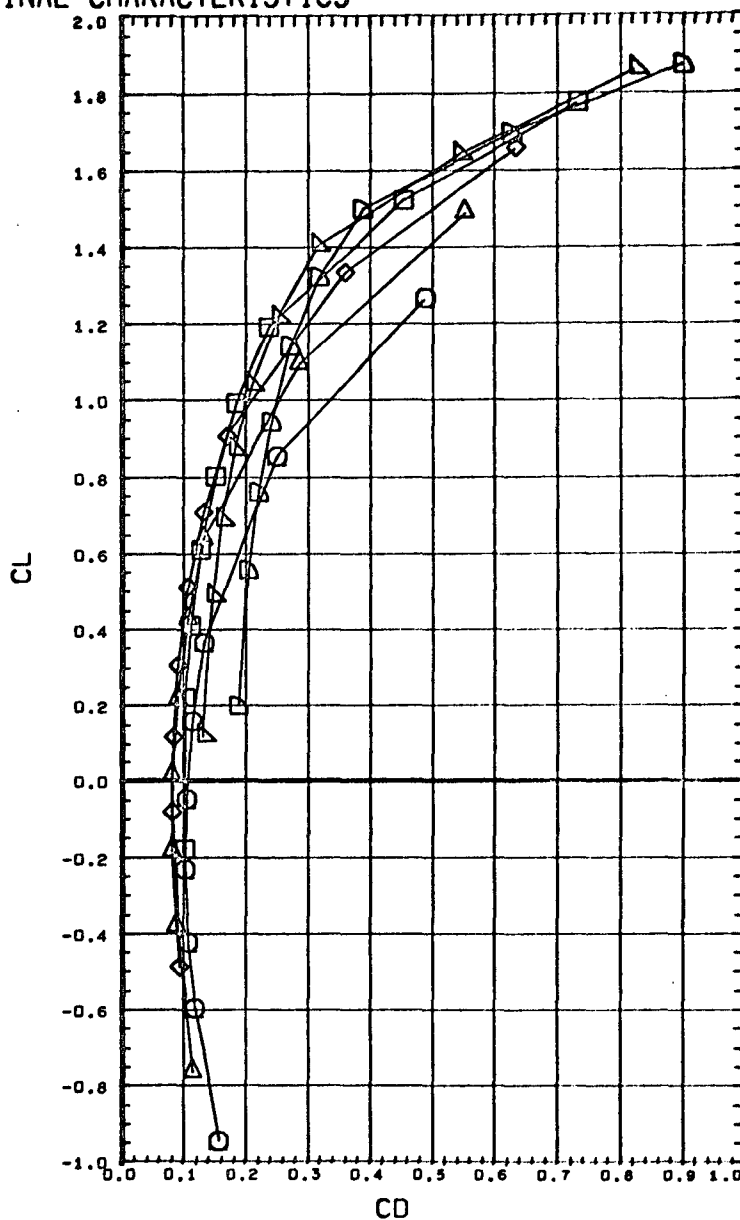
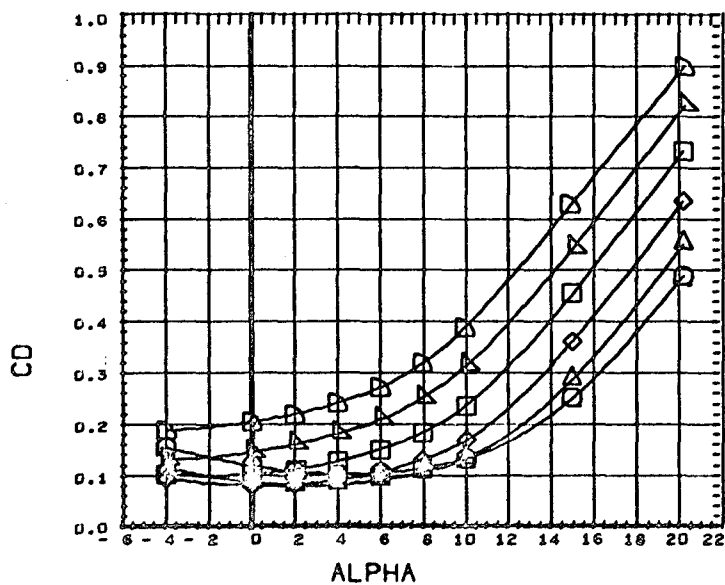
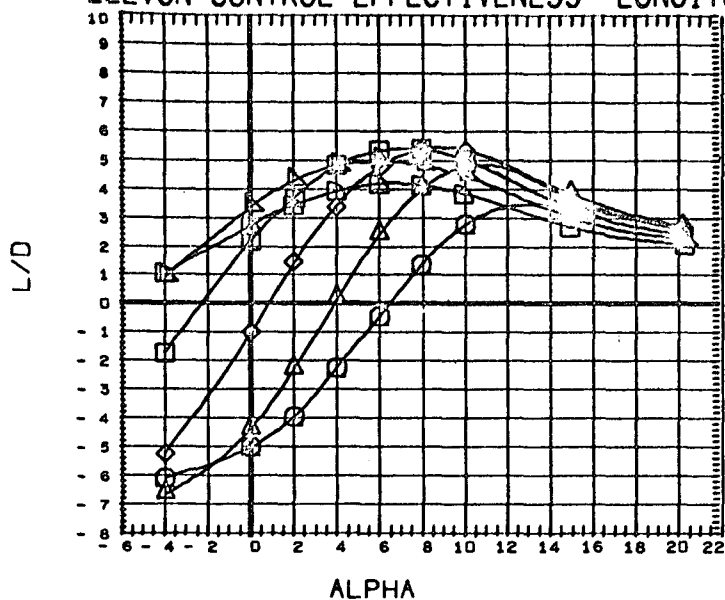
DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RU9D53)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(TU9D17)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(CU9D01)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9D13)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9D21)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9D25)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH 0.398



BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	-20.000	0.000	0.000	SREF 1.3550 80.FT.
0.000	-10.000	0.000	0.000	LREF 3.4530 FT.
0.000	0.000	0.000	0.000	BREF 3.4530 FT.
0.000	10.000	0.000	0.000	XMRF 2.5950 FT.
0.000	20.000	0.000	0.000	YMRF 0.0000 FT.
0.000	30.000	0.000	0.000	ZMRF 0.0187 FT.
				SCALE 1.5000 PER CT

ELEVON CONTROL EFFECTIVENESS- LONGITUDINAL CHARACTERISTICS

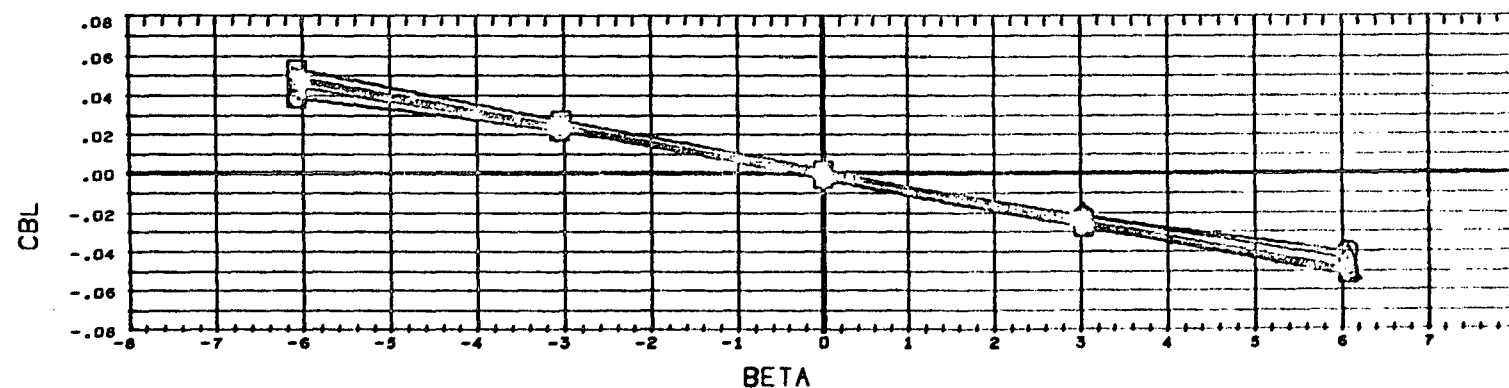
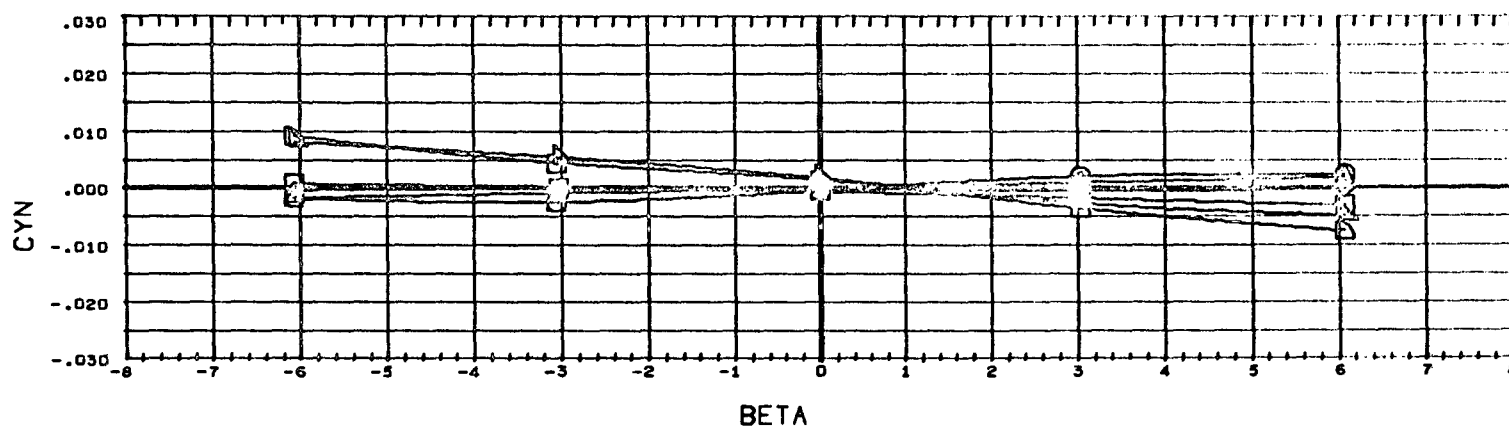
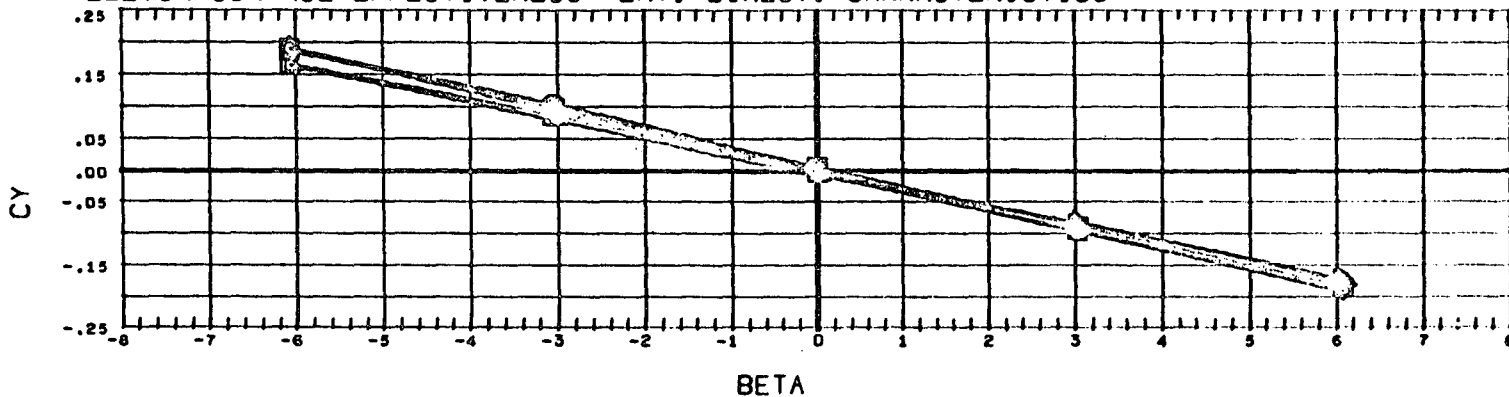


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RU9053)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(TU9017)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(CU9001)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9013)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9021)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9025)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH 0.398

BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	-20.000	0.000	0.000	SREF 1.3550 SQ.FT.
0.000	-10.000	0.000	0.000	LREF 3.4530 FT.
0.000	0.000	0.000	0.000	BREF 3.4530 FT.
0.000	10.000	0.000	0.000	XMRP 2.5950 FT.
0.000	20.000	0.000	0.000	YMRP 0.0000 FT.
0.000	30.000	0.000	0.000	ZMRP 0.0187 FT.
				SCALE 1.5000 PER CT

ELEVON CONTROL EFFECTIVENESS- LAT.-DIRECT. CHARACTERISTICS



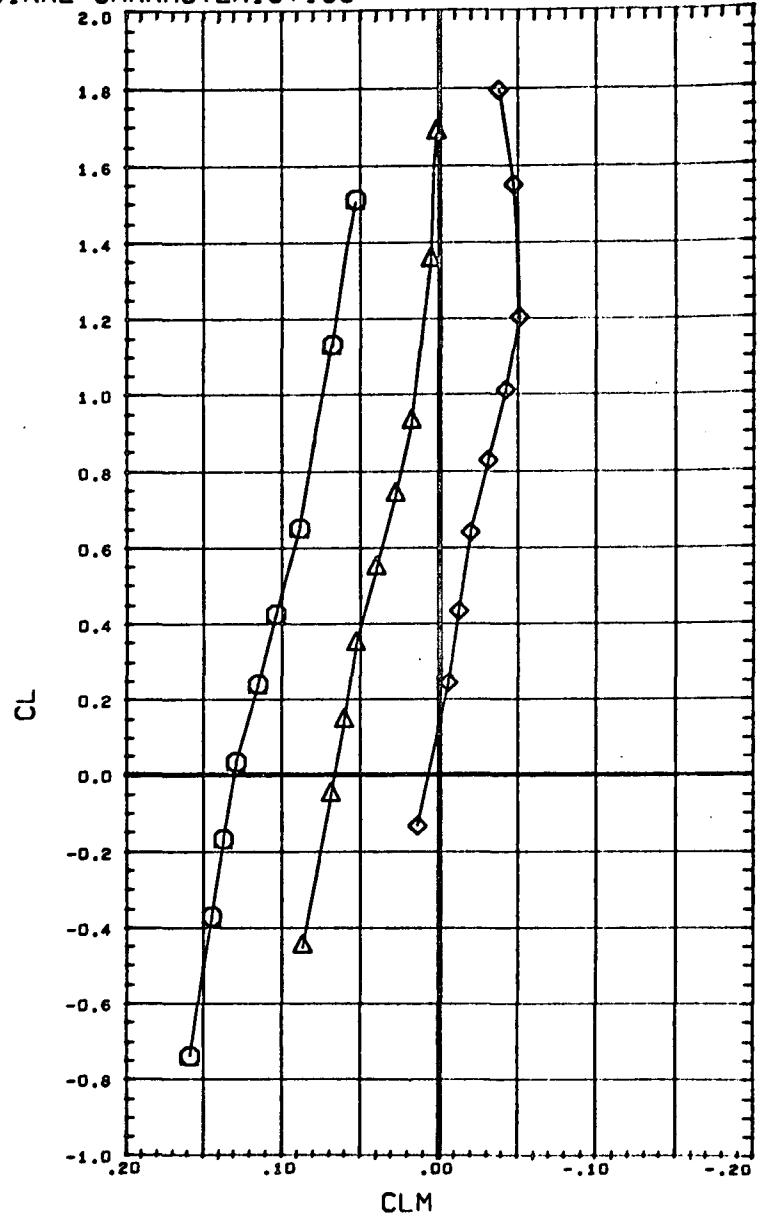
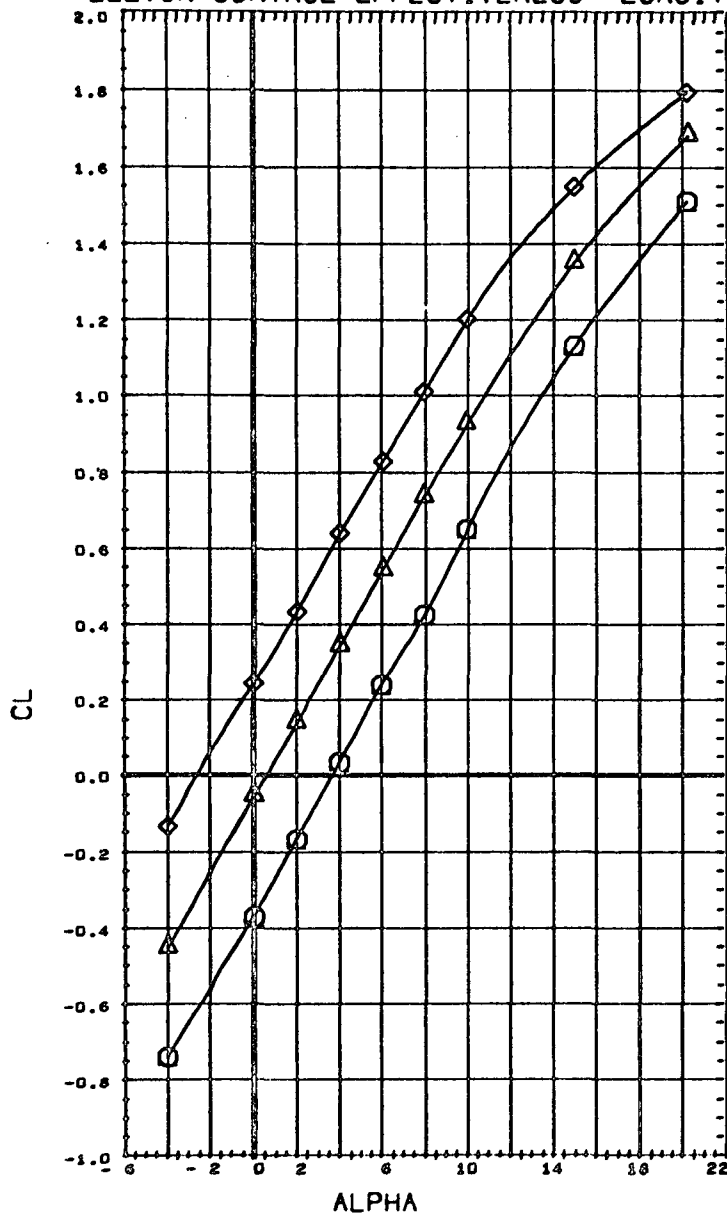
DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RU9054)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(TU9018)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(CU9002)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9014)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9022)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9026)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH

0.399

ALPHA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	-20.000	0.000	0.000	SREF 1.3550 SQ.FT.
0.000	-10.000	0.000	0.000	LREF 3.4530 FT.
0.000	0.000	0.000	0.000	BREF 3.4530 FT.
0.000	10.000	0.000	0.000	XMRF 2.5950 FT.
0.000	20.000	0.000	0.000	YMRF 0.0000 FT.
0.000	30.000	0.000	0.000	ZMRF 0.0187 FT.
				SCALE 1.5000 PER CT

ELEVON CONTROL EFFECTIVENESS- LONGITUDINAL CHARACTERISTICS

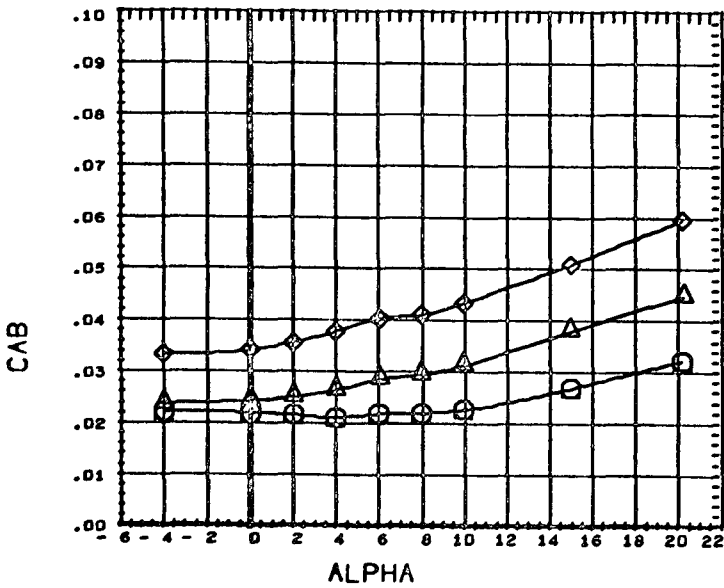
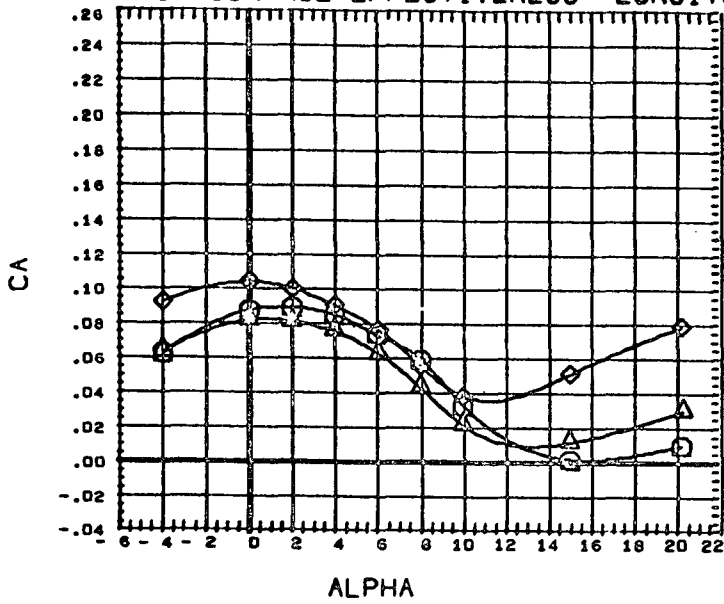


DATA SET SYMBOL CONFIGURATION DESCRIPTION
(RU9049) ○ CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9029) △ CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9045) ◇ CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH 0.399

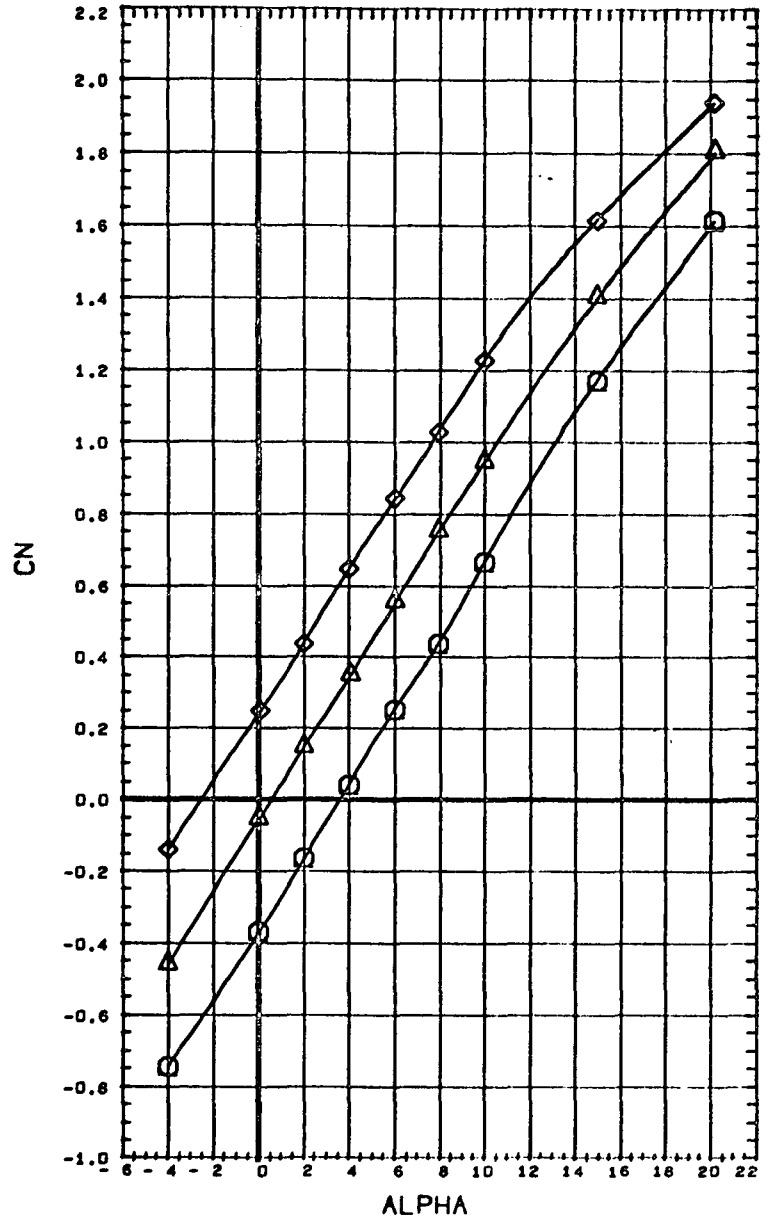
BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	-10.000	10.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	10.000	0.000	LREF 3.4530 FT.
0.000	10.000	10.000	0.000	BREF 3.4530 FT.
				XMRP 2.5950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0187 FT.
				SCALE 1.5000 PER CT

ELEVON CONTROL EFFECTIVENESS- LONGITUDINAL CHARACTERISTICS



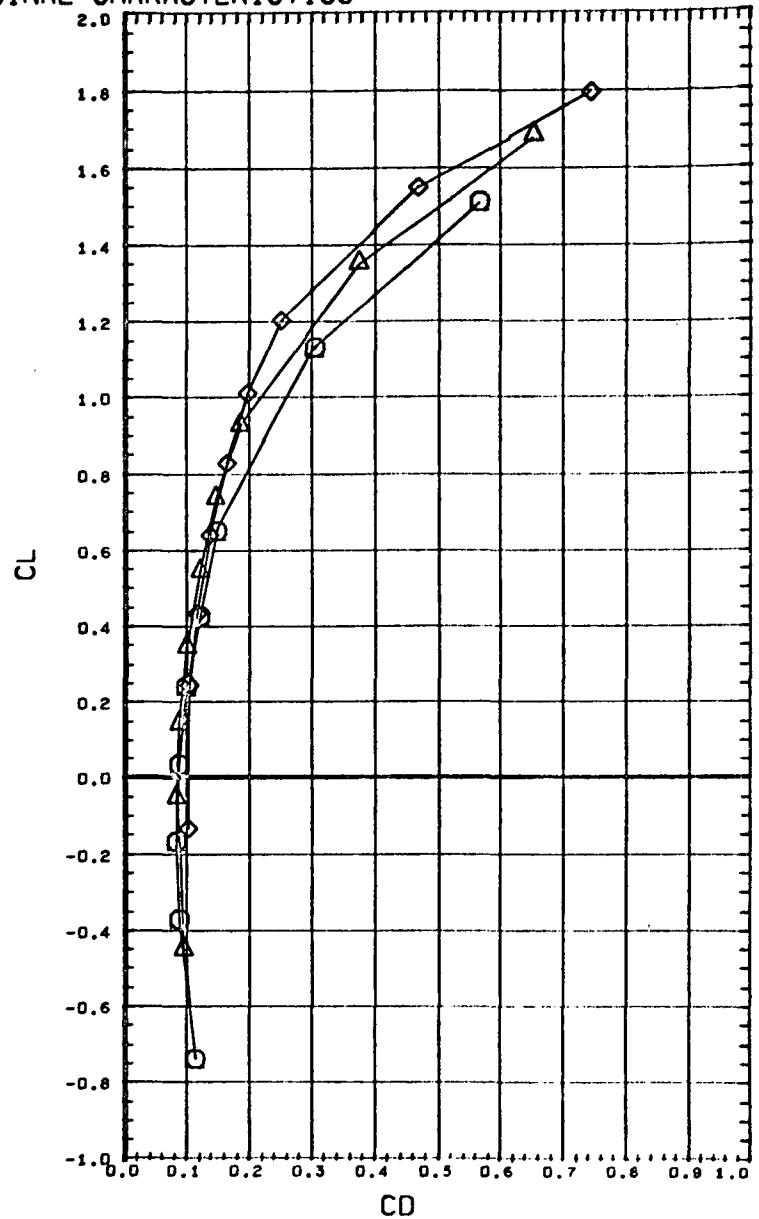
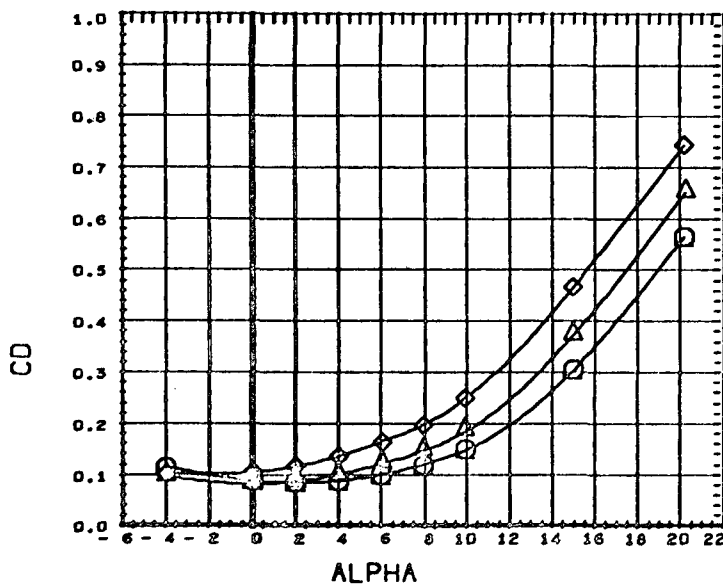
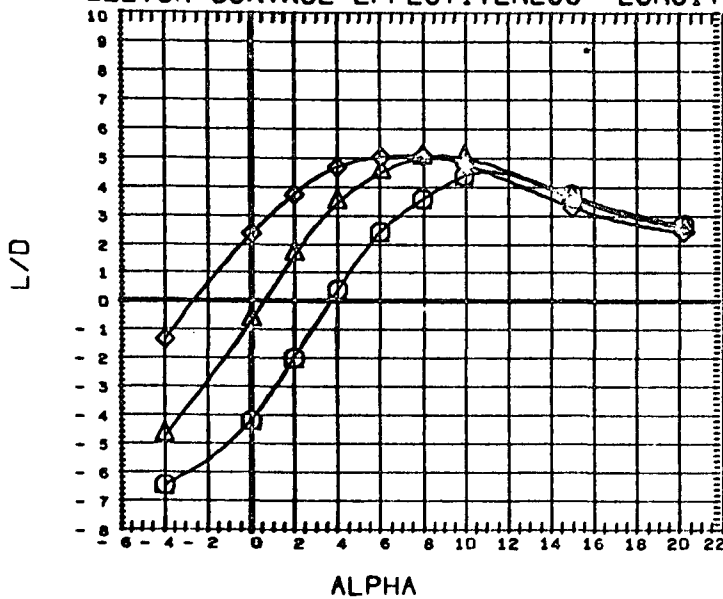
DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RU9049)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9029)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9045)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH 0.399



BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	-10.000	10.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	10.000	0.000	LREF 3.4530 FT.
0.000	10.000	10.000	0.000	BREF 3.4530 FT.
				XMRP 2.5950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0187 FT.
				SCALE 1.5000 PER CT

ELEVON CONTROL EFFECTIVENESS- LONGITUDINAL CHARACTERISTICS

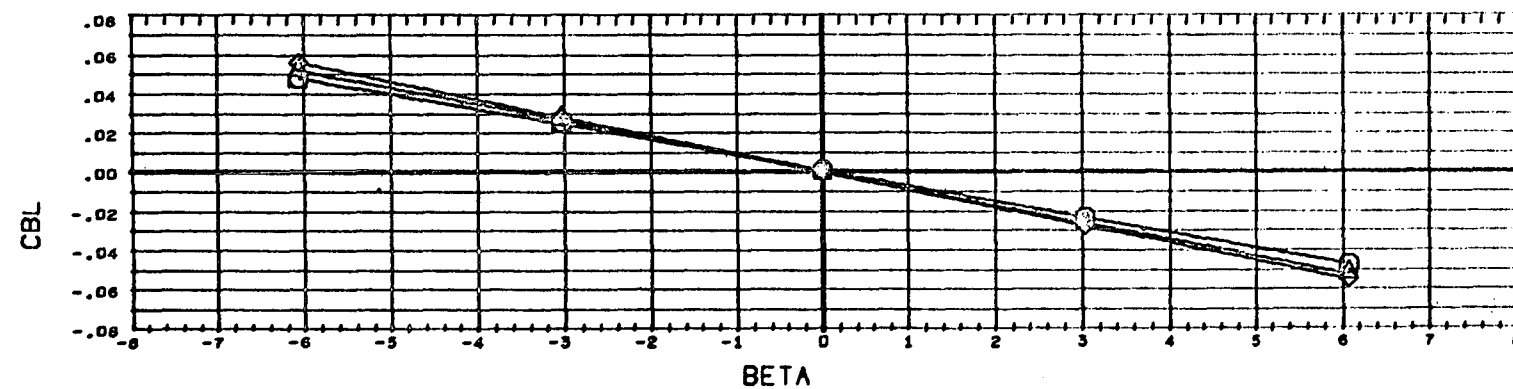
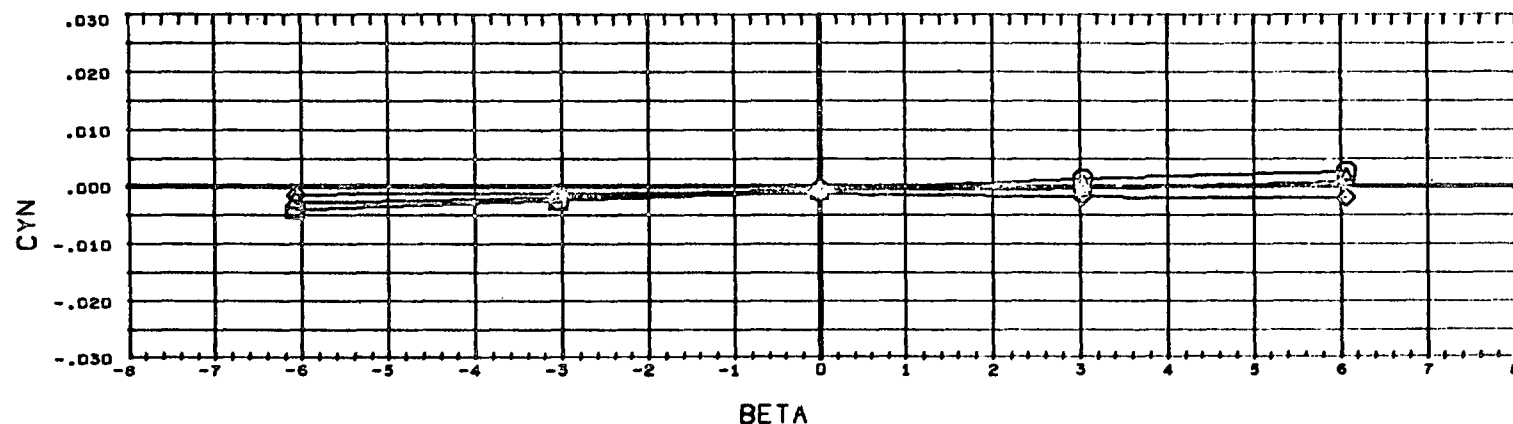
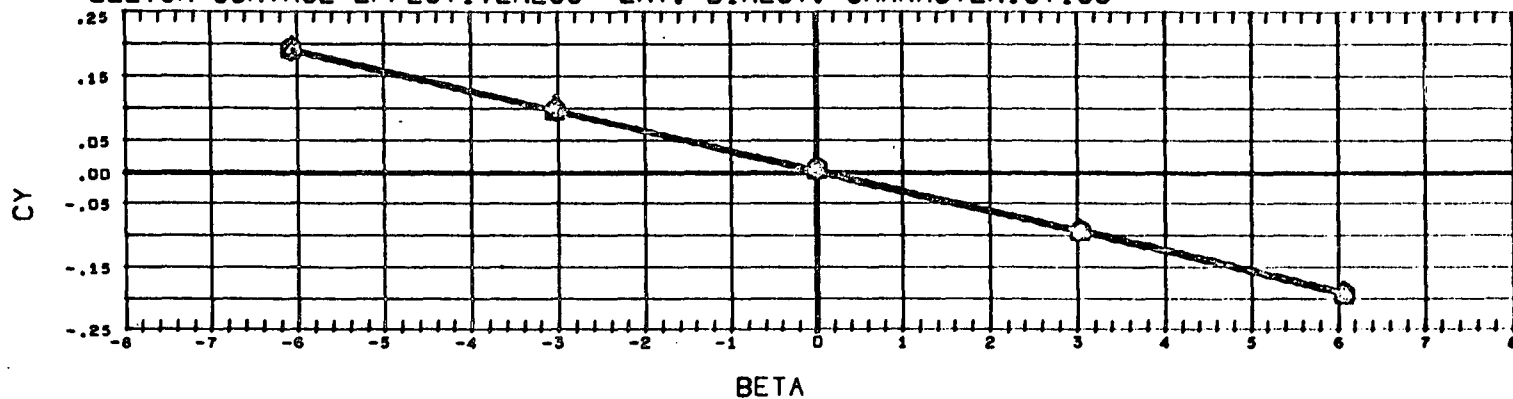


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RU9049)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9029)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9045)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH 0.399

BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	-10.000	10.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	10.000	0.000	LREF 3.4530 FT.
0.000	10.000	10.000	0.000	BREF 3.4530 FT.
				XMRP 2.5950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0187 FT.
				SCALE 1.5000 PER CY

ELEVON CONTROL EFFECTIVENESS- LAT.-DIRECT. CHARACTERISTICS

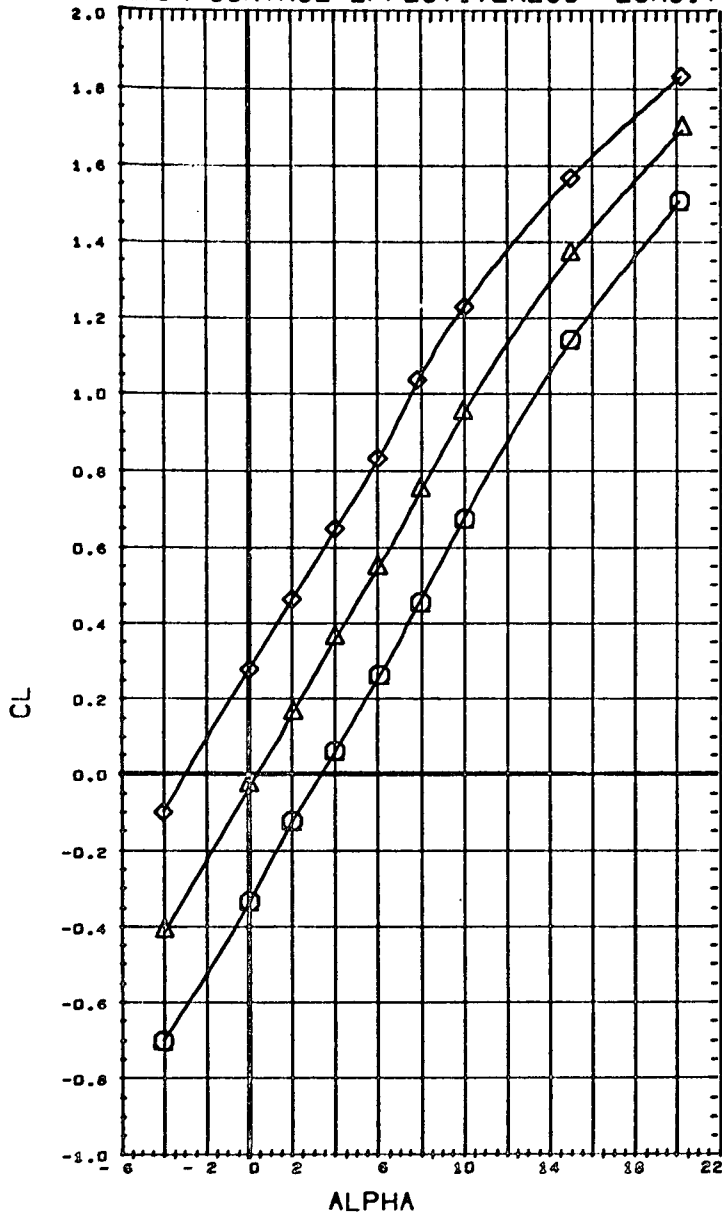


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RU9D55)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9D35)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9D46)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1

ALPHA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	-10.000	10.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	10.000	0.000	LREF 3.4530 FT.
0.000	10.000	10.000	0.000	BREF 3.4530 FT.
				XMRP 2.5950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0167 FT.
				SCALE 1.5000 PER CT

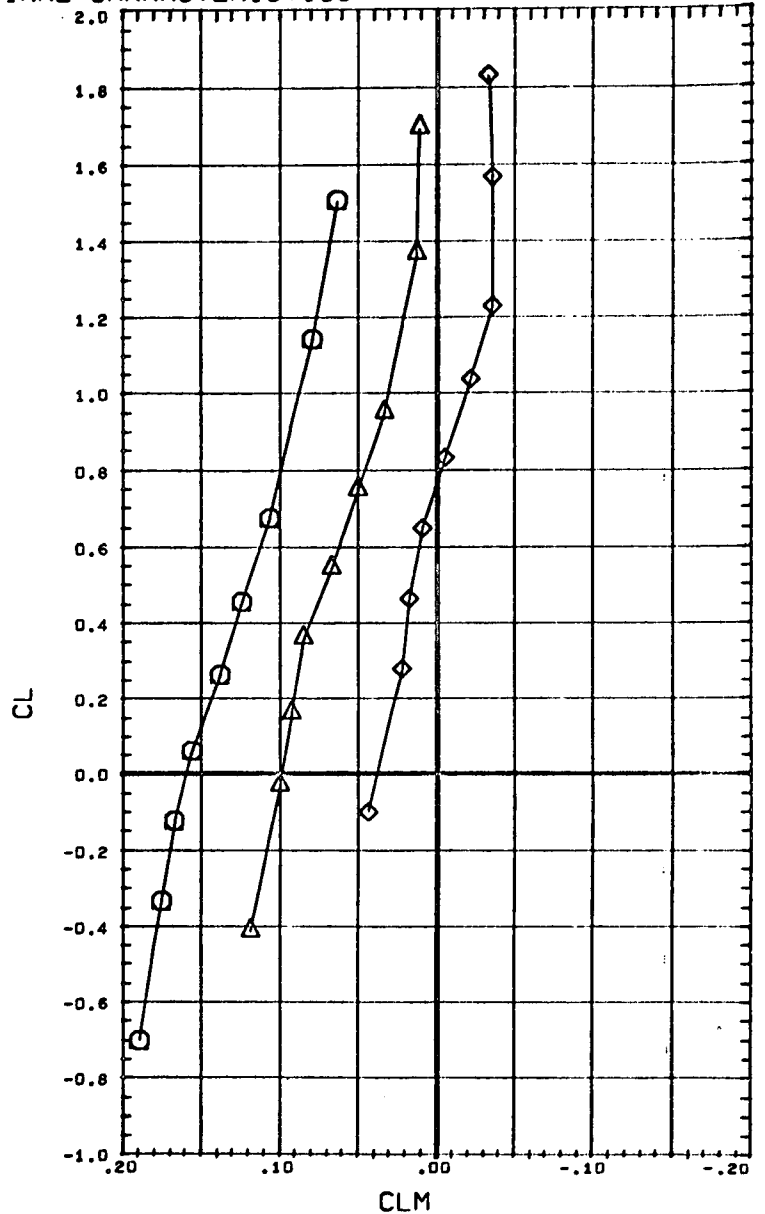
MACH 0.398

ELEVON CONTROL EFFECTIVENESS- LONGITUDINAL CHARACTERISTICS



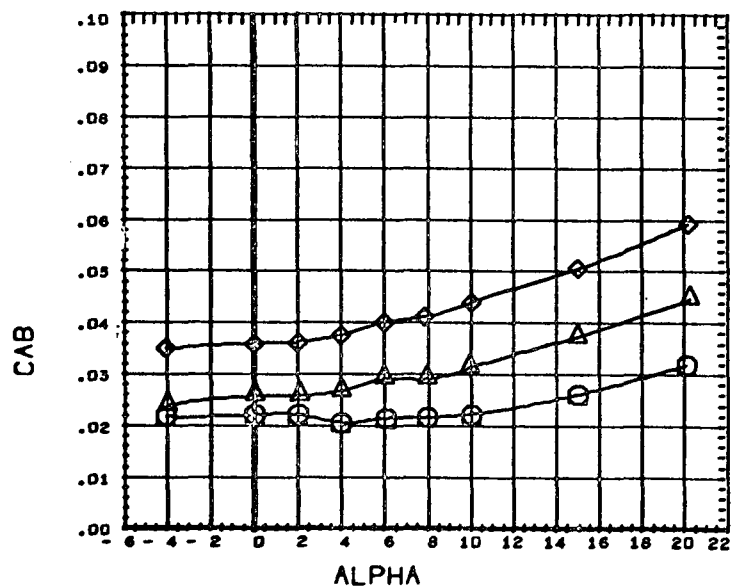
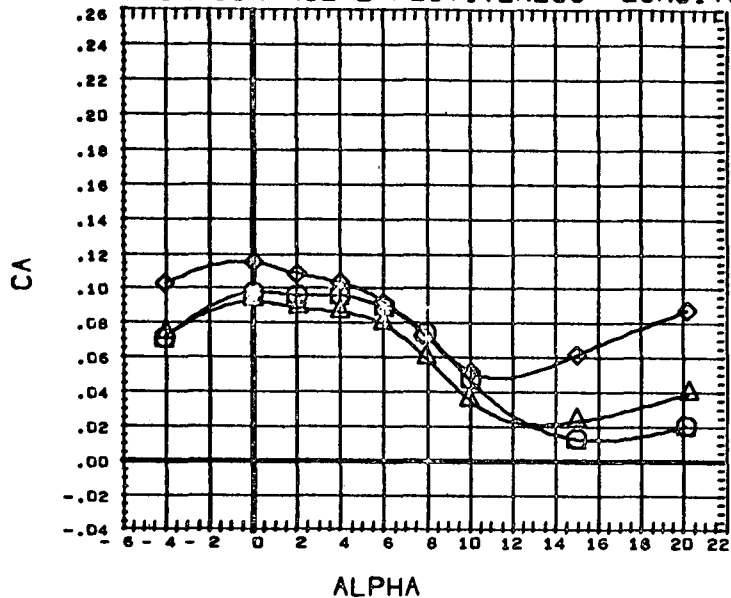
DATA SET SYMBOL CONFIGURATION DESCRIPTION
(RU9D41) (Circle) CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9D33) (Triangle) CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9D37) (Diamond) CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH 0.398



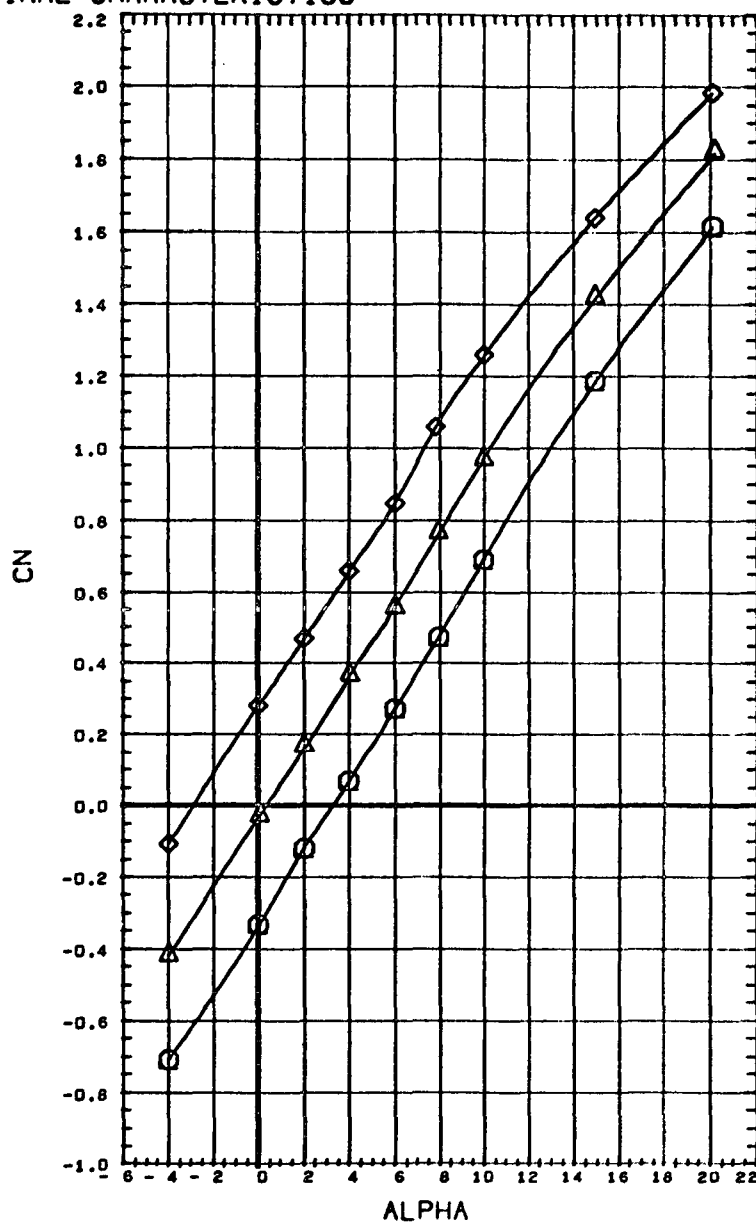
BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION	
0.000	-10.000	20.000	0.000	SREF	1.3550 SQ.FT.
0.000	0.000	20.000	0.000	LREF	3.4530 FT.
0.000	10.000	20.000	0.000	BREF	3.4530 FT.
				XMRF	2.5950 FT.
				YMRF	0.0000 FT.
				ZMRF	0.0187 FT.
				SCALE	1.5000 PER CT

ELEVON CONTROL EFFECTIVENESS- LONGITUDINAL CHARACTERISTICS



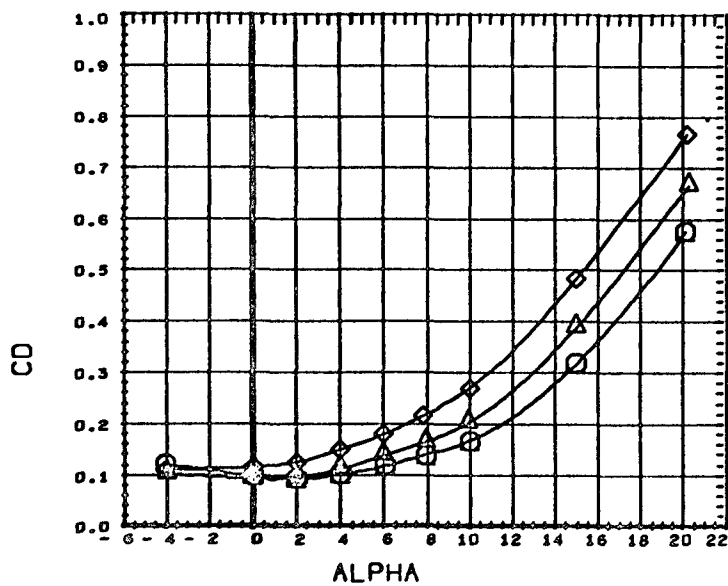
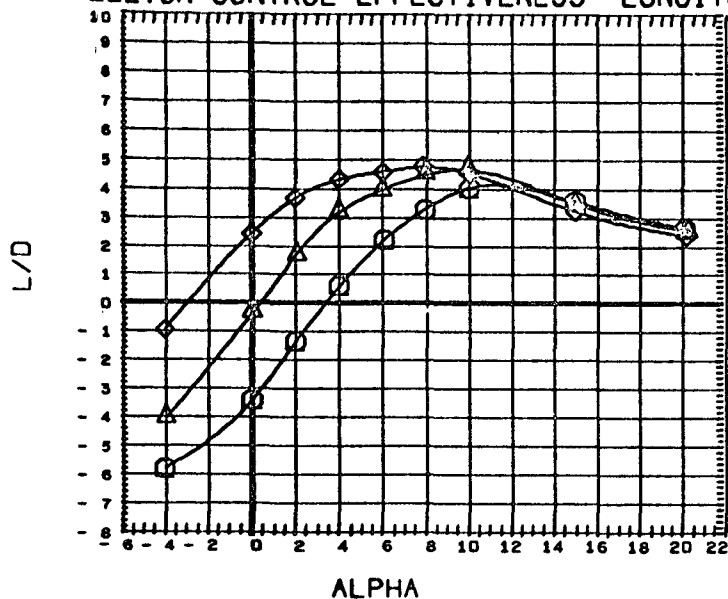
DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RU9041)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9035)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9037)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH 0.398



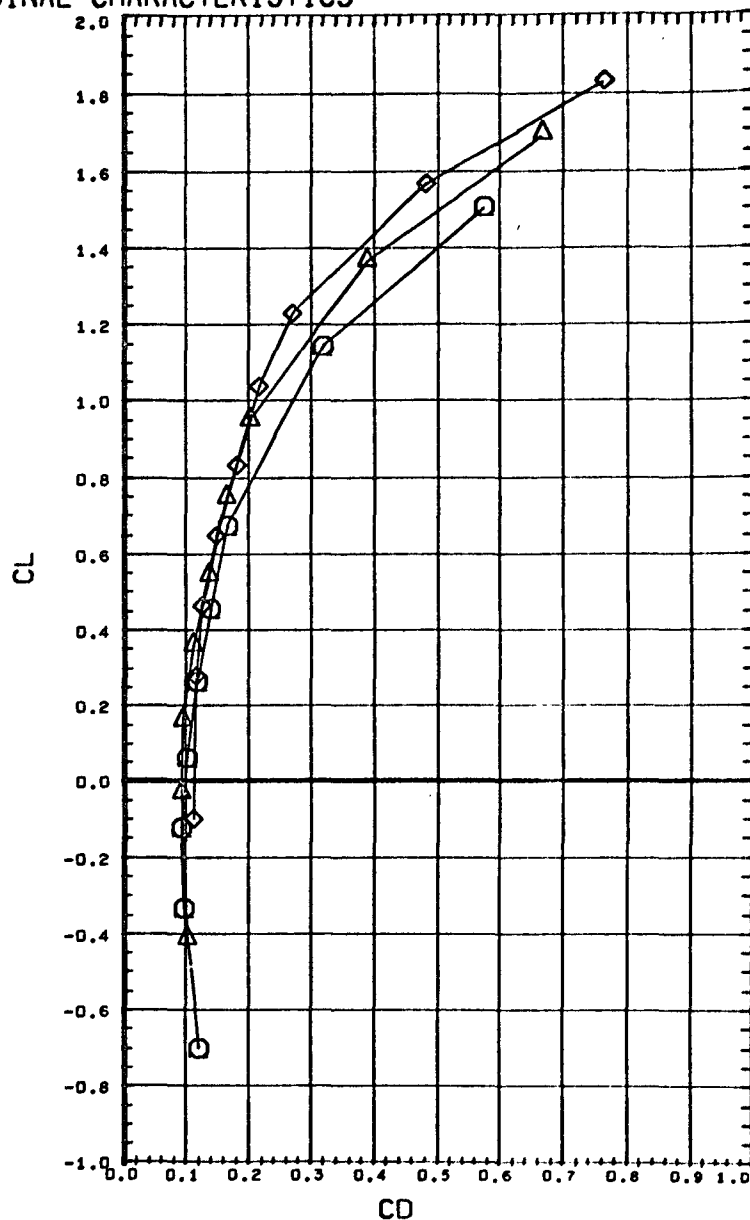
BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	-10.000	20.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	20.000	0.000	LREF 3.4530 FT.
0.000	10.000	20.000	0.000	BREF 3.4530 FT.
				XMRP 2.5950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0187 FT.
				SCALE 1.9000 PER CT

ELEVON CONTROL EFFECTIVENESS- LONGITUDINAL CHARACTERISTICS



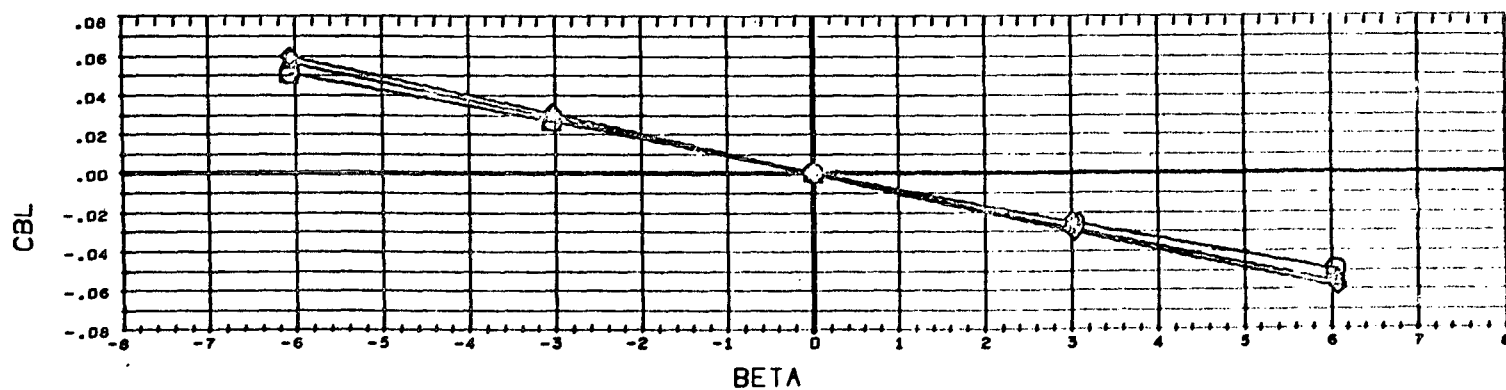
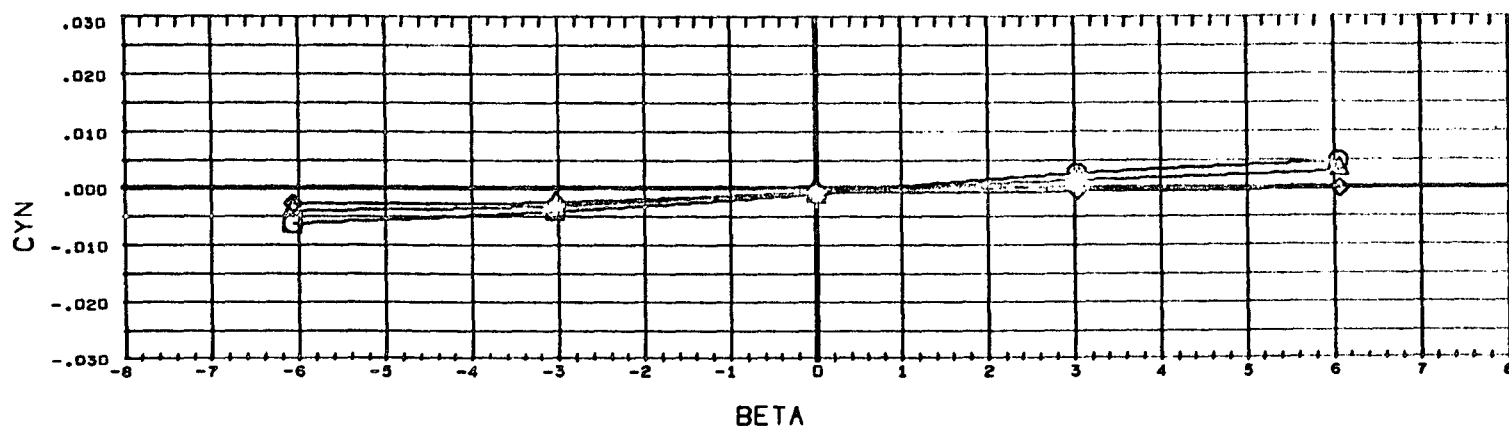
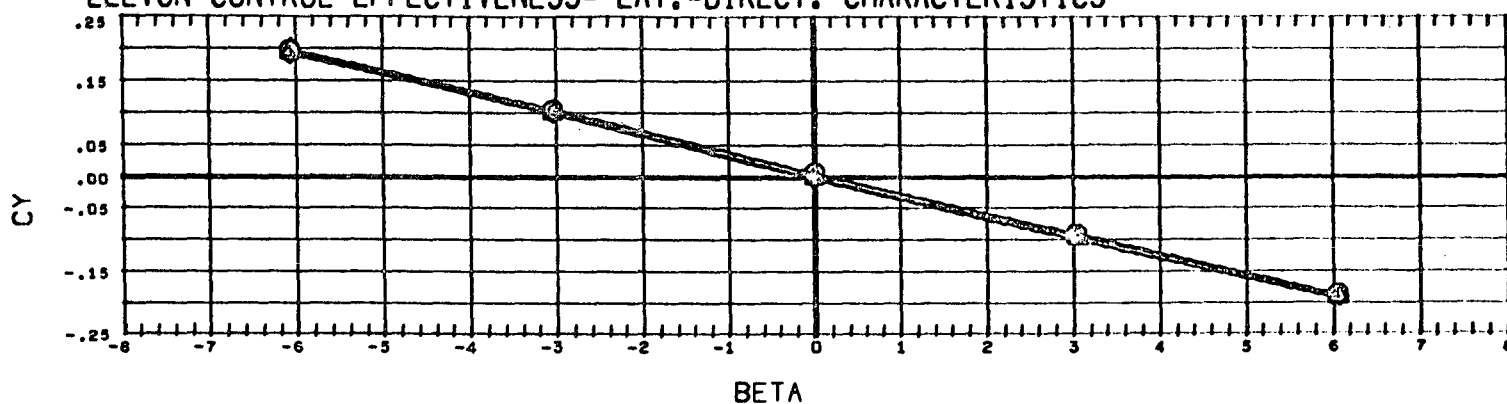
DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RU9341)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9333)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9337)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACM 0.398



BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	-10.000	20.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	20.000	0.000	LREF 3.4530 FT.
0.000	10.000	20.000	0.000	BREF 3.4530 FT.
				XMRP 2.5950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0187 FT.
				SCALE 1.5000 PER CT

ELEVON CONTROL EFFECTIVENESS- LAT.-DIRECT. CHARACTERISTICS

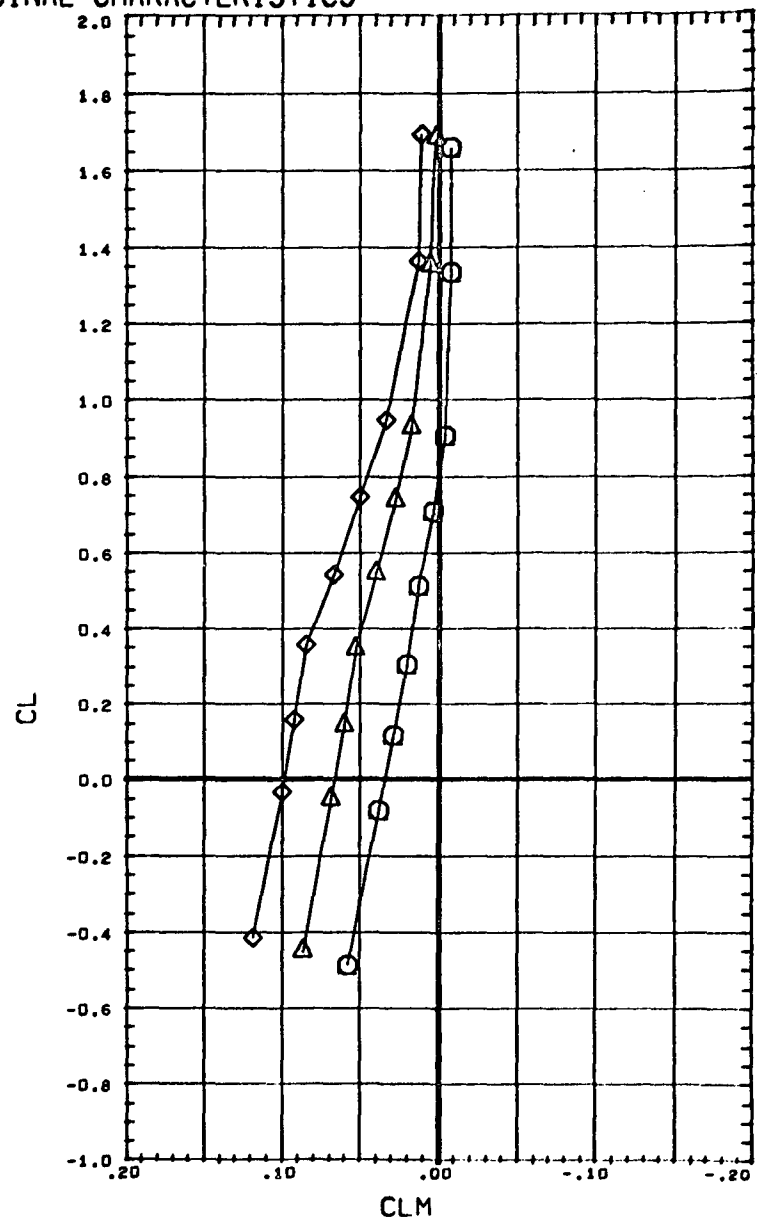
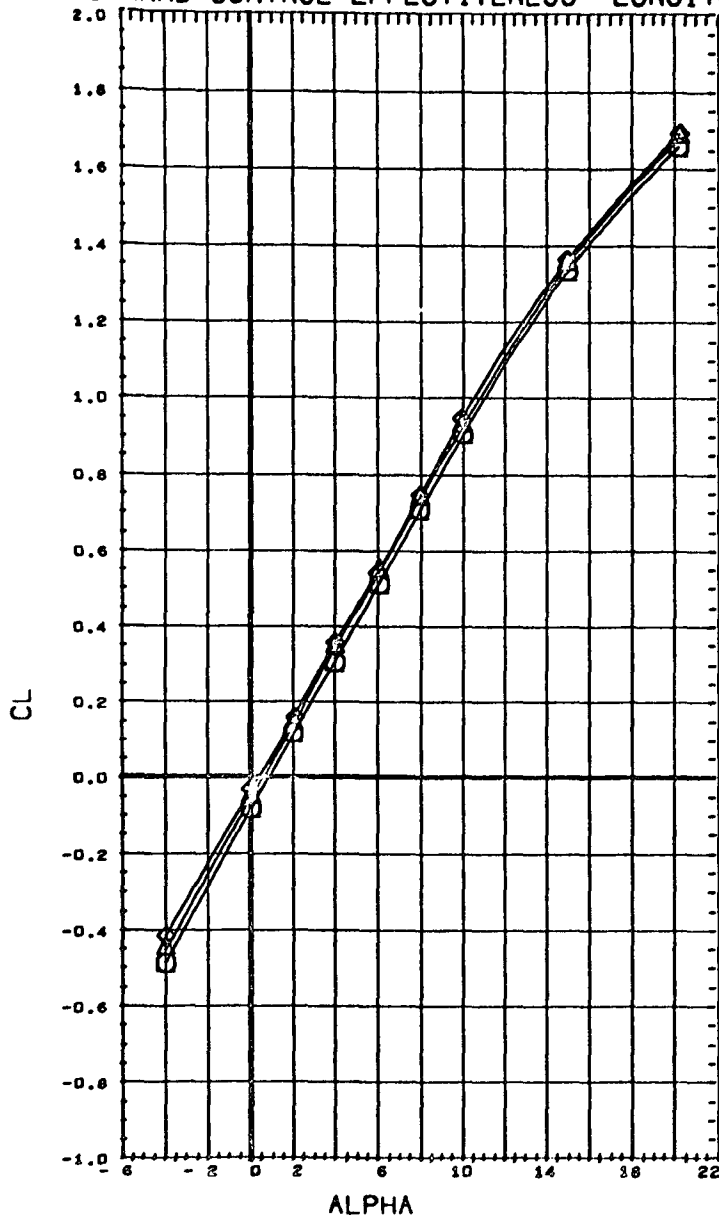


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RU9042)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9034)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9038)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1

ALPHA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	-10.000	20.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	20.000	0.000	LREF 3.4530 FT.
0.000	10.000	20.000	0.000	BREF 3.4530 FT.
				XMRP 2.5950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0187 FT.
				SCALE 1.5000 PER CT

MACH 0.399

CANARD CONTROL EFFECTIVENESS- LONGITUDINAL CHARACTERISTICS

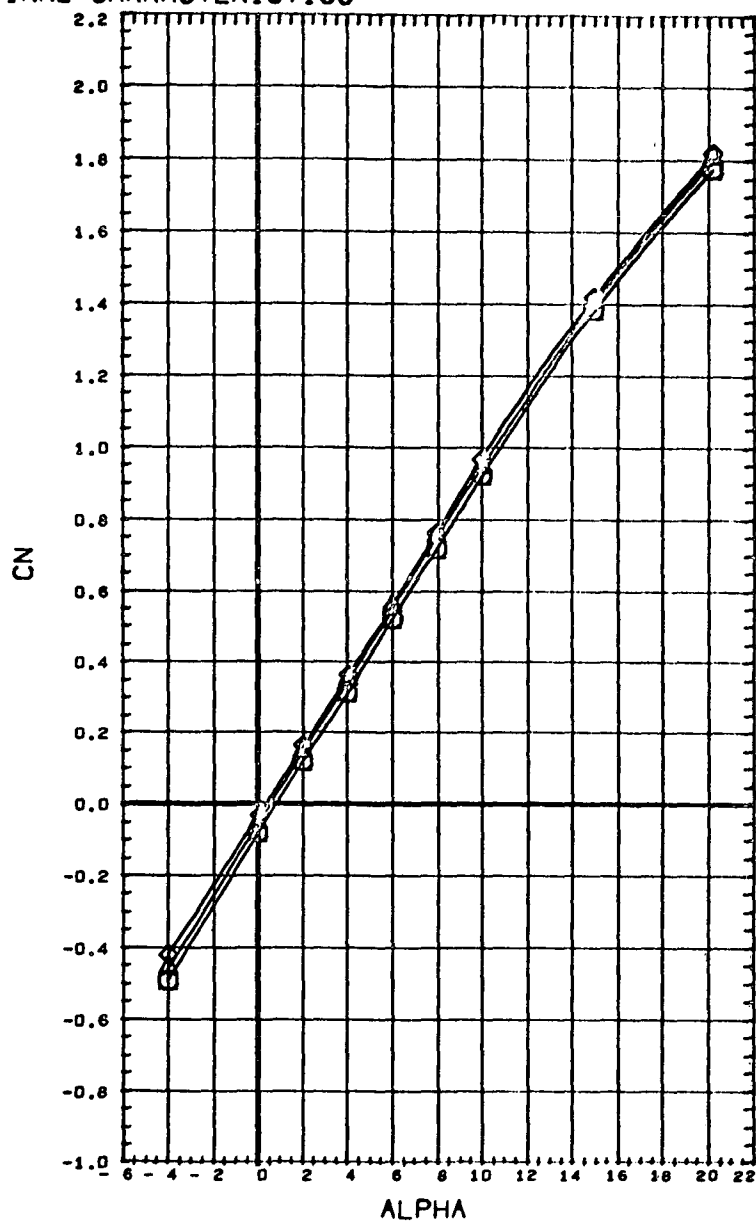
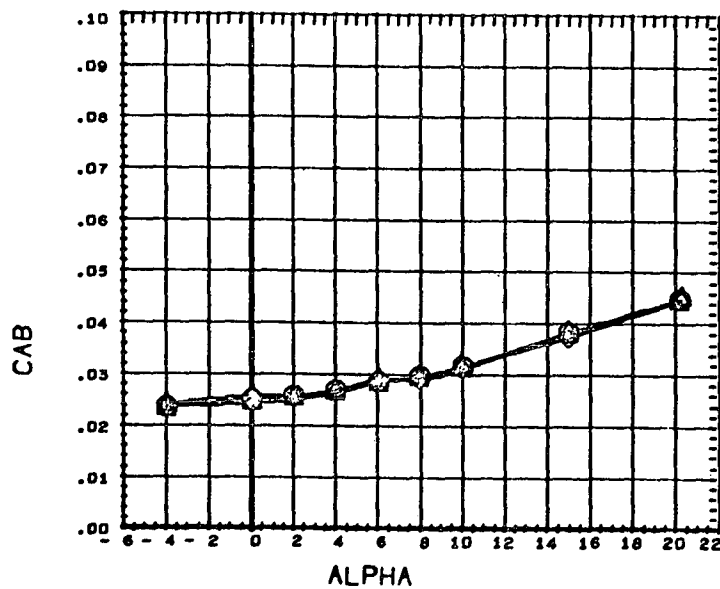
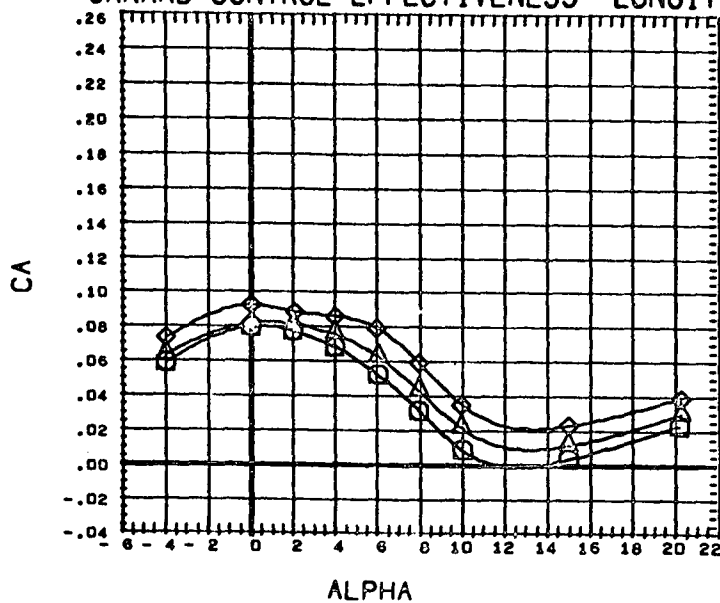


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(CU9001)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9029)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9033)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH 0.400

BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	10.000	0.000	LREF 3.4530 FT.
0.000	0.000	20.000	0.000	BREF 3.4530 FT.
				XMRP 2.5950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0187 FT.
				SCALE 1.5000 PER CT

CANARD CONTROL EFFECTIVENESS- LONGITUDINAL CHARACTERISTICS

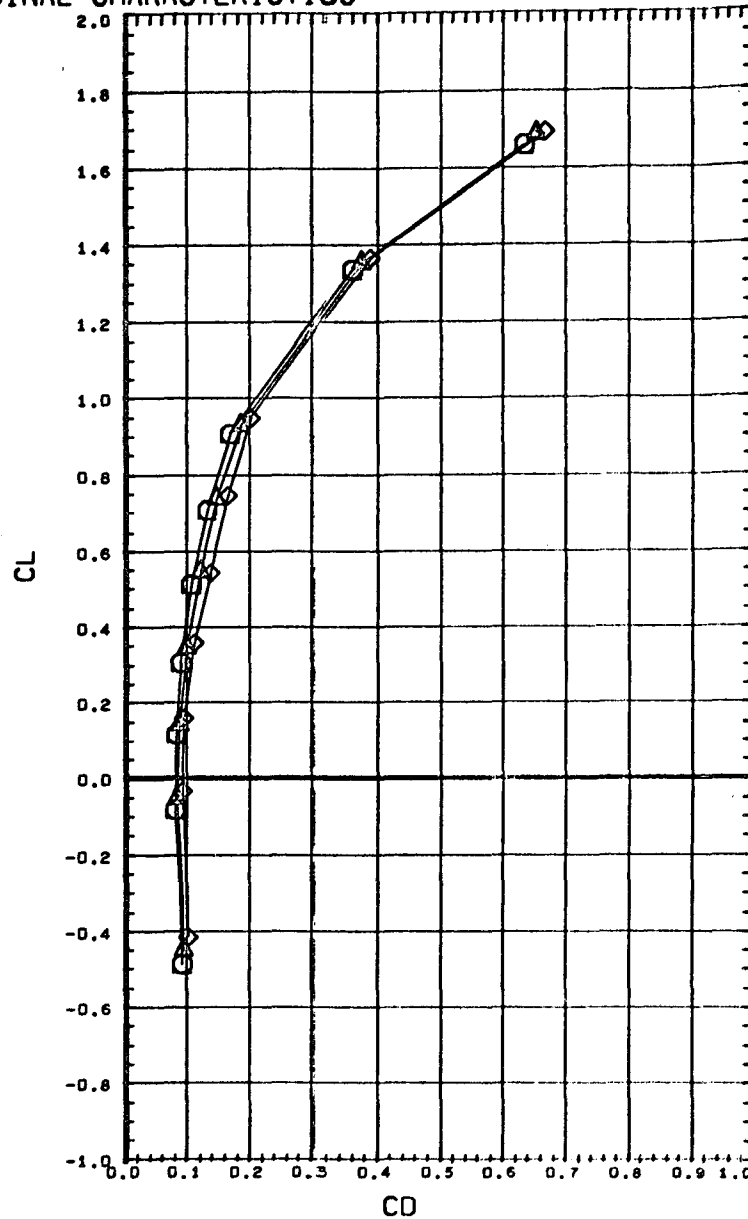
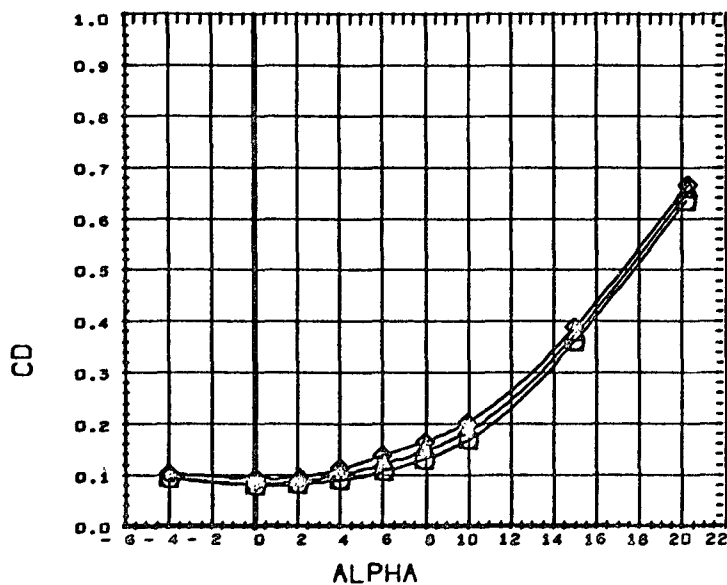
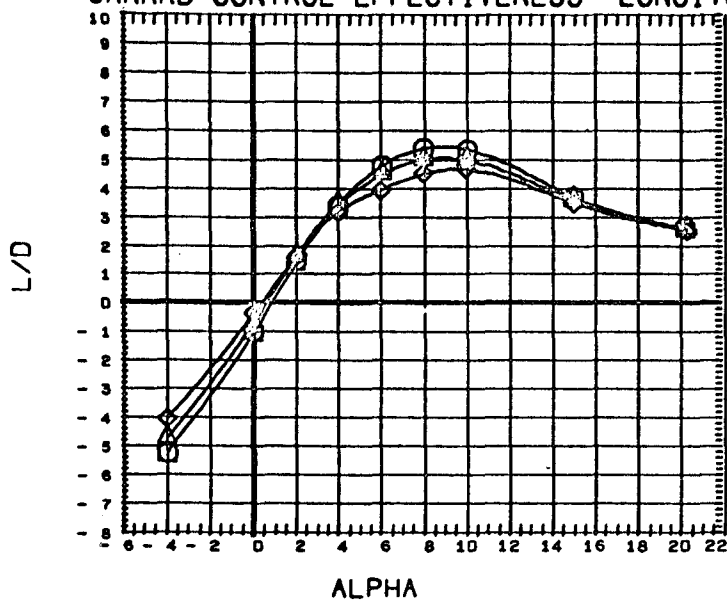


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(CU9001)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9029)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9033)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH 0.400

BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	10.000	0.000	LREF 3.4530 FT.
0.000	0.000	20.000	0.000	BREF 3.4530 FT.
				XMRP 2.9950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0167 FT.
				SCALE 1.5000 PER CT

CANARD CONTROL EFFECTIVENESS- LONGITUDINAL CHARACTERISTICS

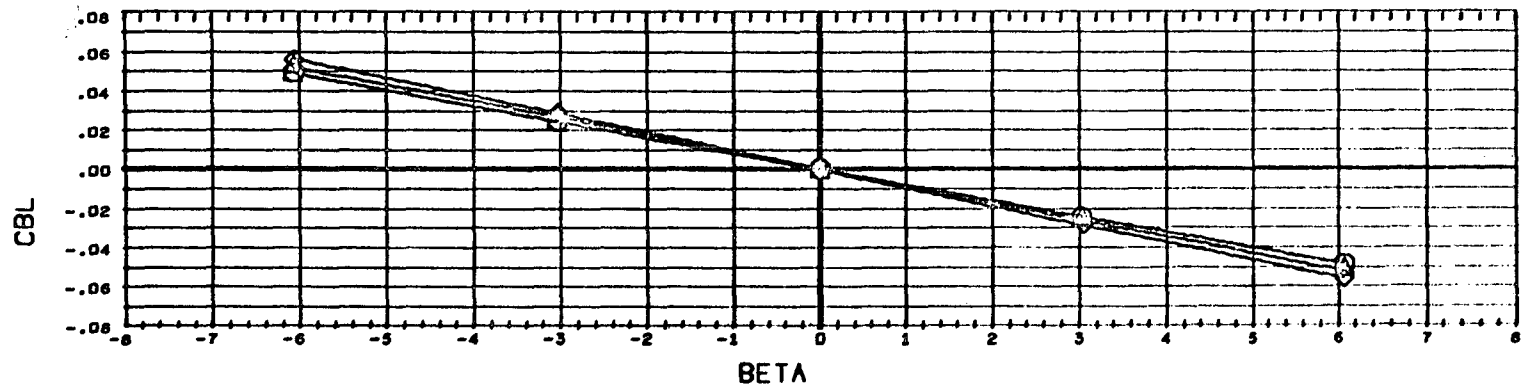
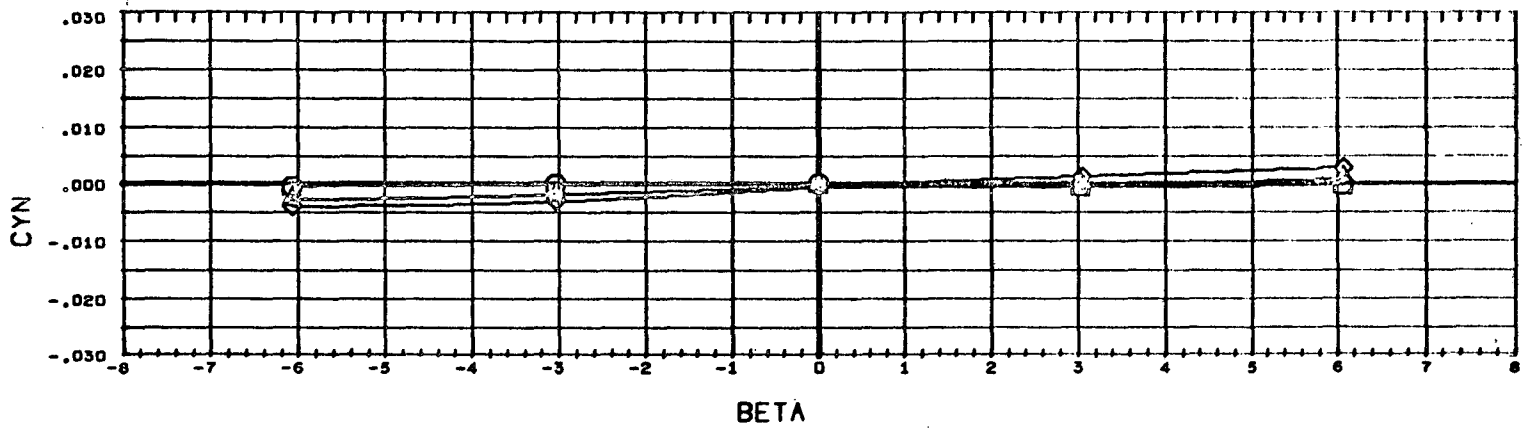
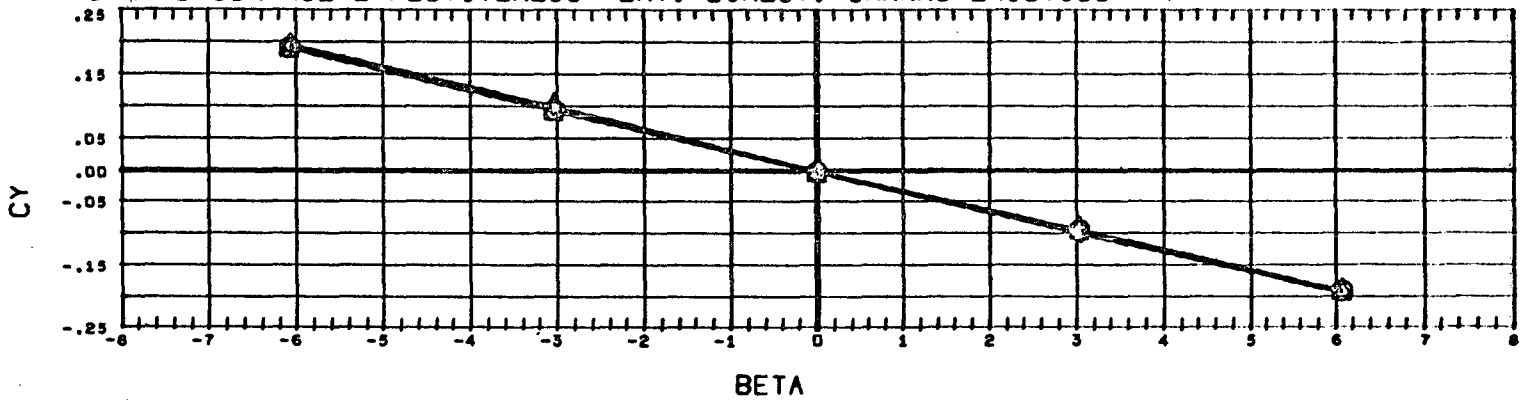


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(CU9001)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9029)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9033)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1

MACH 0.400

BETA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	10.000	0.000	LREF 3.4530 FT.
0.000	0.000	20.000	0.000	BREF 3.4530 FT.
				XMRF 2.5950 FT.
				YMRF 0.0000 FT.
				ZMRF 0.0187 FT.
				SCALE 1.5000 PER CT

CANARD CONTROL EFFECTIVENESS- LAT.-DIRECT. CHARACTERISTICS

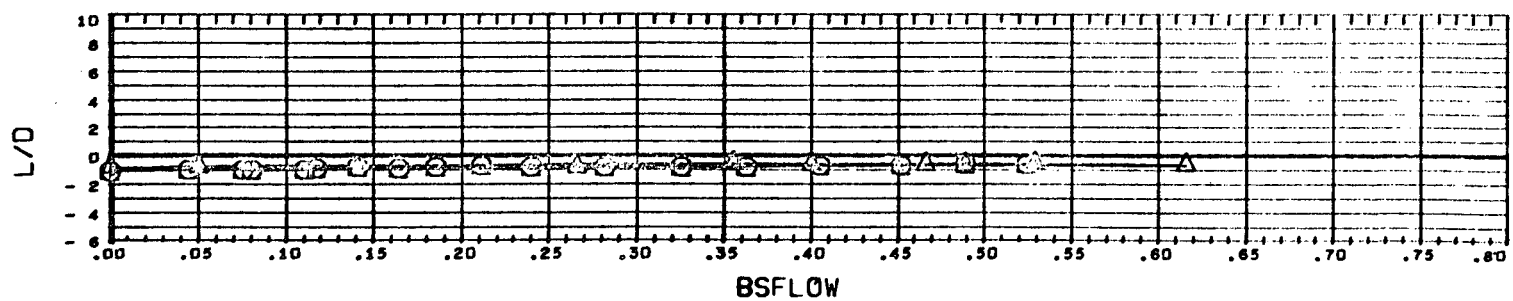
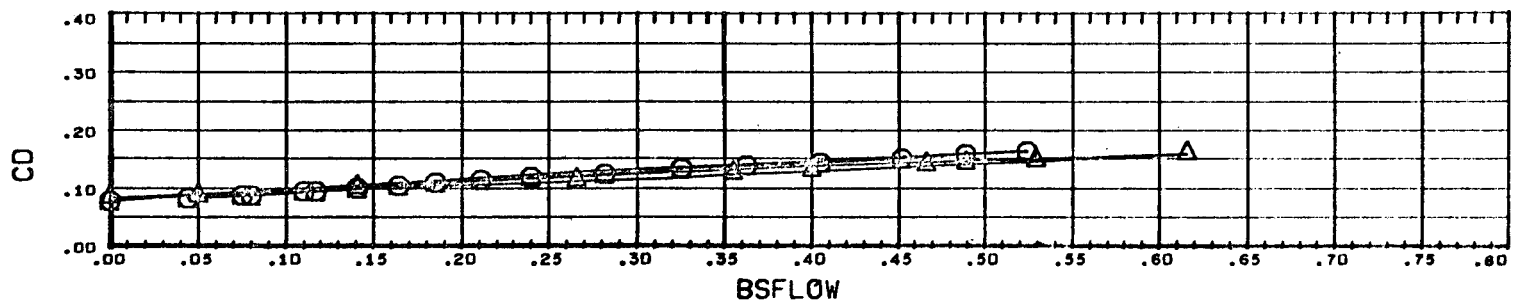
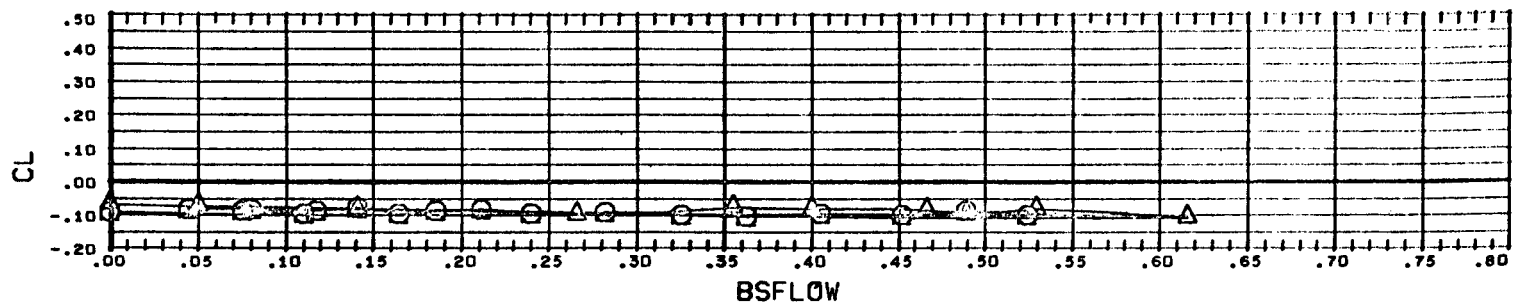
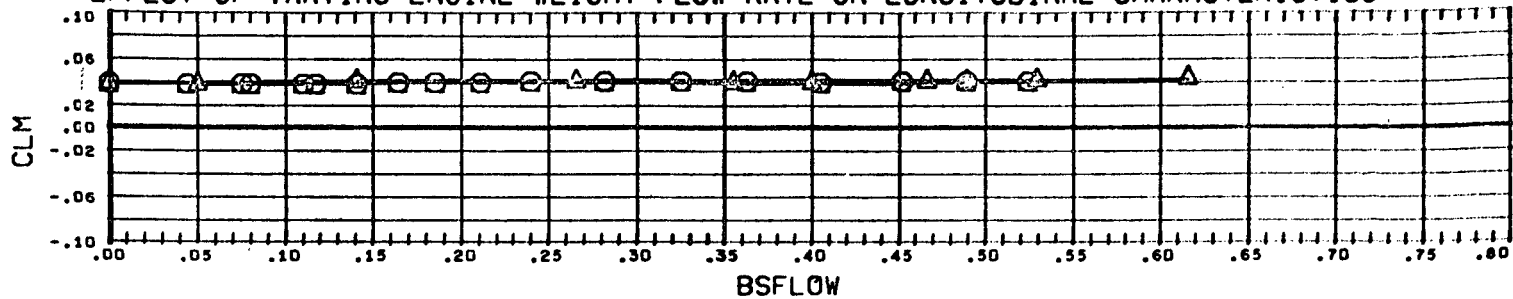


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(CU9002)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9030)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9034)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1

ALPHA	ELEVTR	CANARD	BSFLOW	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 1.3550 80.FT.
0.000	0.000	10.000	0.000	LREF 3.4530 FT.
0.000	0.000	20.000	0.000	BREF 3.4530 FT.
				XMRP 2.5950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0187 FT.
				SCALE 1.5000 PER CT

MACH 0.399

EFFECT OF VARYING ENGINE WEIGHT FLOW RATE ON LONGITUDINAL CHARACTERISTICS

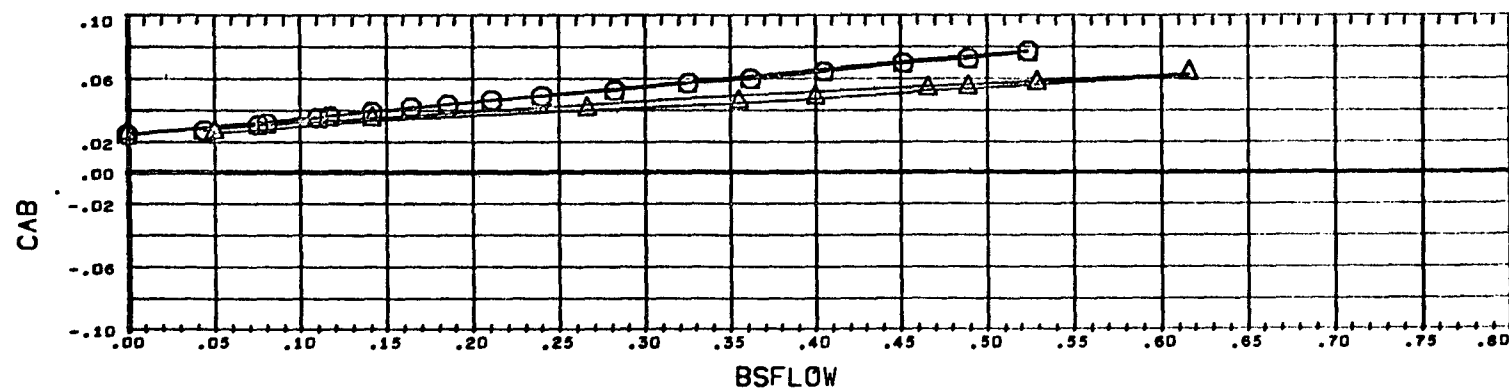
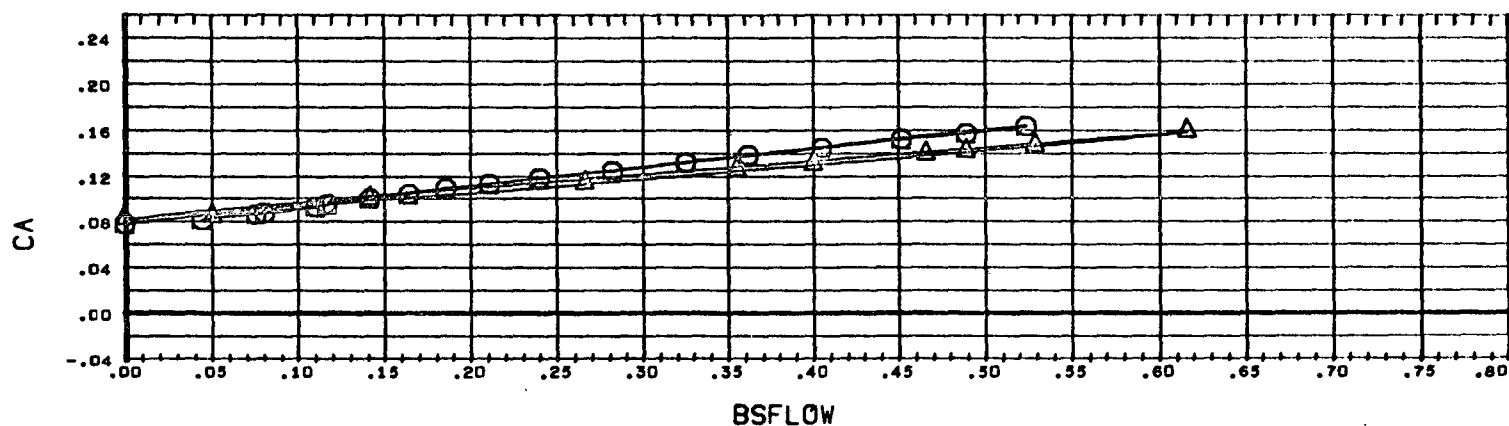
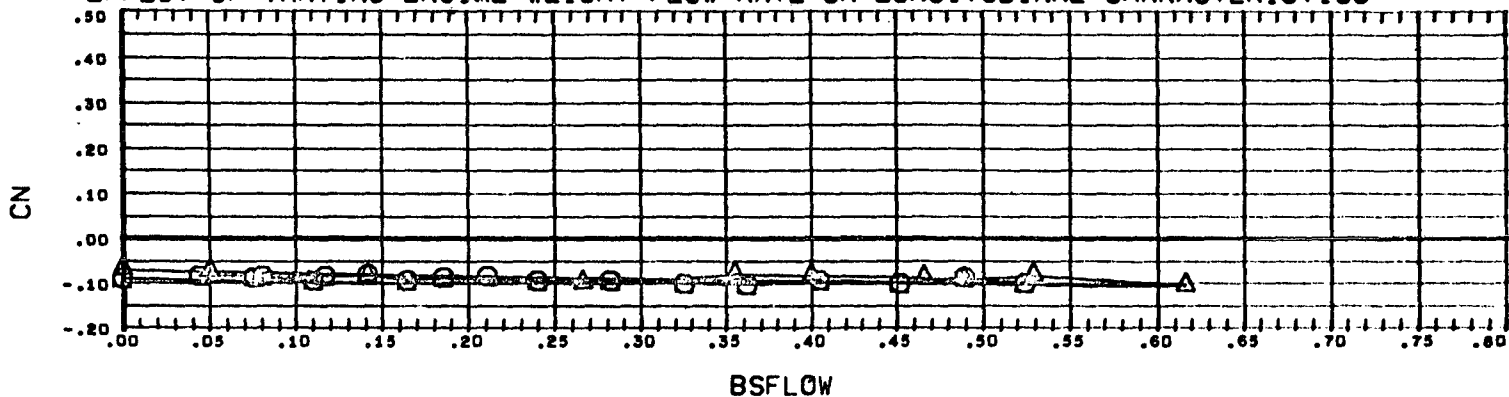


DATA SET SYMBOL	CONFIGURATION DESCRIPTION
(RU9061)	CAL MSFC/LMSC BOOSTER B4C2F2W3V1
(RU9066)	CAL MSFC/LMSC BOOSTER B5C2F2W3V1

ALPHA	BETA	ELEVTR	CANARD	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 1.3550 SQ.FT.
0.000	0.000	0.000	0.000	LREF 3.4530 FT.
				BREF 3.4530 FT.
				XMHP 2.5950 FT.
				YMHP 0.0000 FT.
				ZMHP 0.0137 FT.
				SCALE 1.5000 PER CT

MACH 0.400

EFFECT OF VARYING ENGINE WEIGHT FLOW RATE ON LONGITUDINAL CHARACTERISTICS

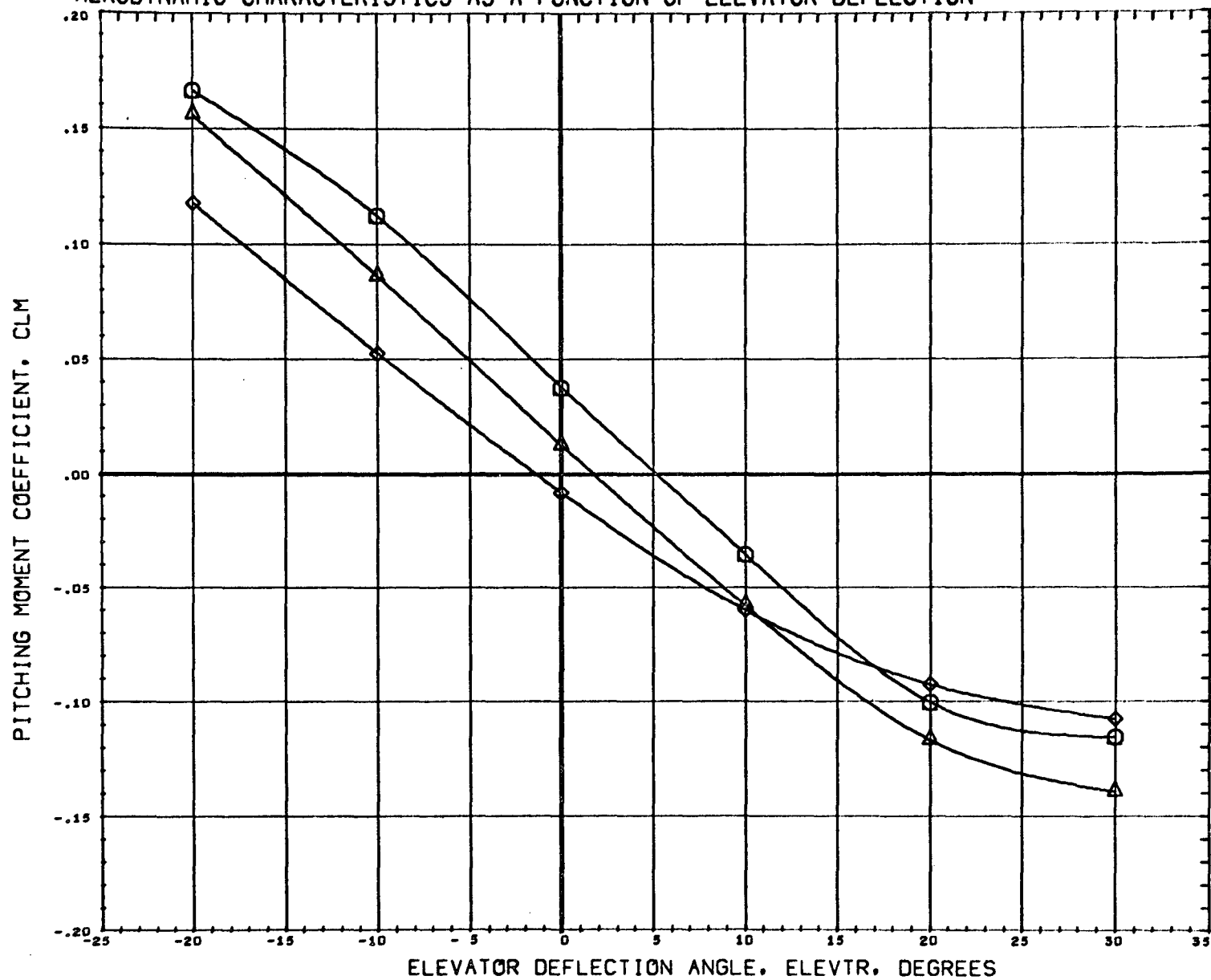


DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (RU9061) CAL MSFC/LMSC BOOSTER B4C2F2W3V1
 (RU9066) CAL MSFC/LMSC BOOSTER B5C2F2W3V1

ALPHA	BETA	ELEVTR	CANARD	REFERENCE INFORMATION
0.000	0.000	0.000	0.000	SREF 1.3550 39.FT.
0.000	0.000	0.000	0.000	LREF 3.4530 FT.
				BREF 3.4530 FT.
				XMRP 2.5950 FT.
				YMRP 0.0000 FT.
				ZMRP 0.0187 FT.
				SCALE 1.5000 PER CT

MACH 0.400

AERODYNAMIC CHARACTERISTICS AS A FUNCTION OF ELEVATOR DEFLECTION



SYMBOL	ALPHA	PARAMETRIC VALUES			
○	0.000	MACH	0.400	BETA	0.000
△	6.000	CANARD	0.000	BSFLOW	0.000
◇	15.000				

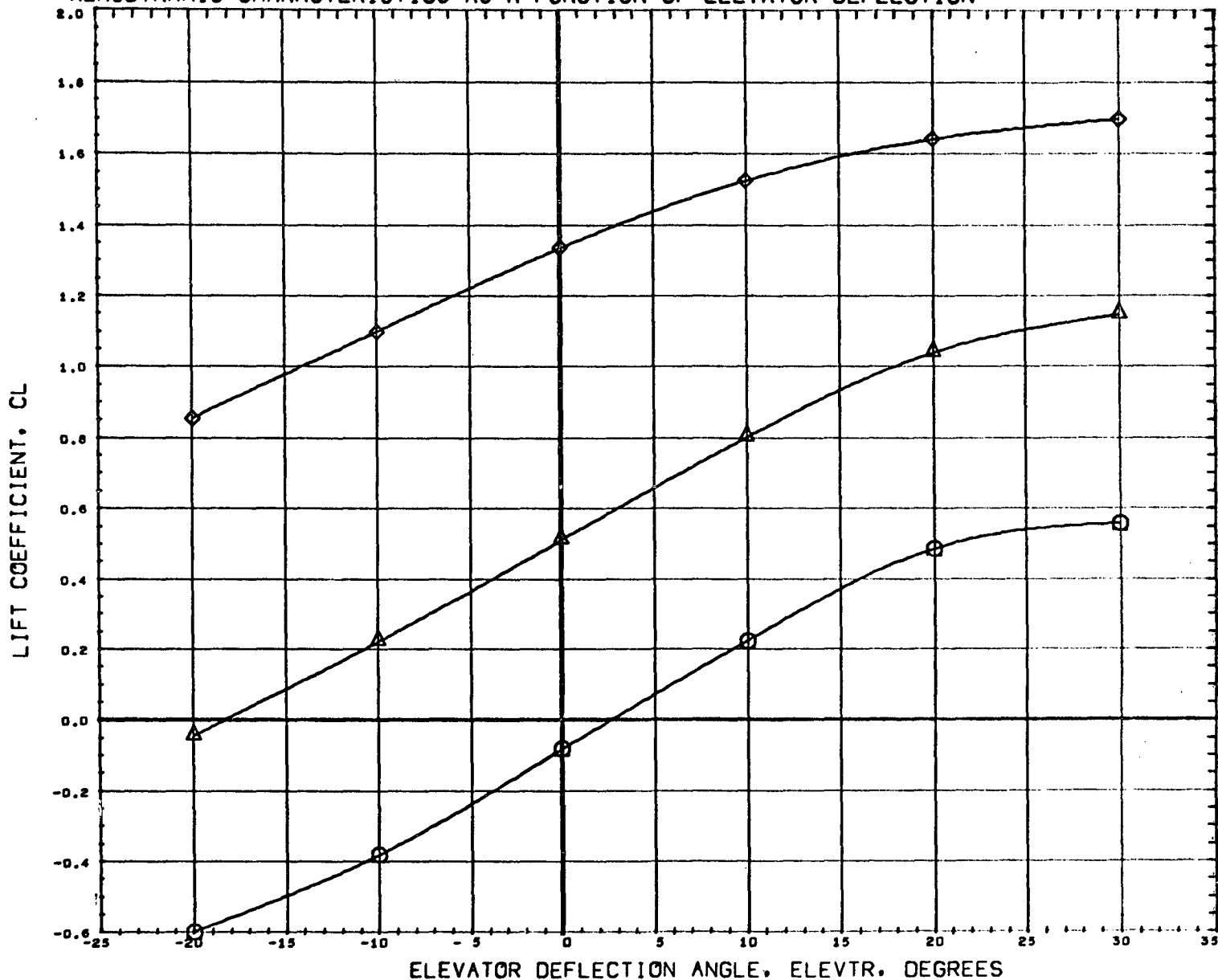
REFERENCE INFORMATION		
SREF	1.3550	80.FT.
LREF	3.4530	FT.
BREF	3.4530	FT.
XMRP	2.5950	FT.
YMRP	0.0000	FT.
ZMRP	0.0187	FT.
SCALE	1.9000	PER CT

DATA HIST. CODE MMI

CAL MSFC/LMSC BOOSTER B4C2F2W3V1

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AERODYNAMIC CHARACTERISTICS AS A FUNCTION OF ELEVATOR DEFLECTION



SYMBOL	ALPHA	MACH	PARAMETRIC VALUES	BETA	0.000
○	0.000	0.400	BSFLOW	0.000	
△	6.000	CANARD	0.000	0.000	
◇	15.000				

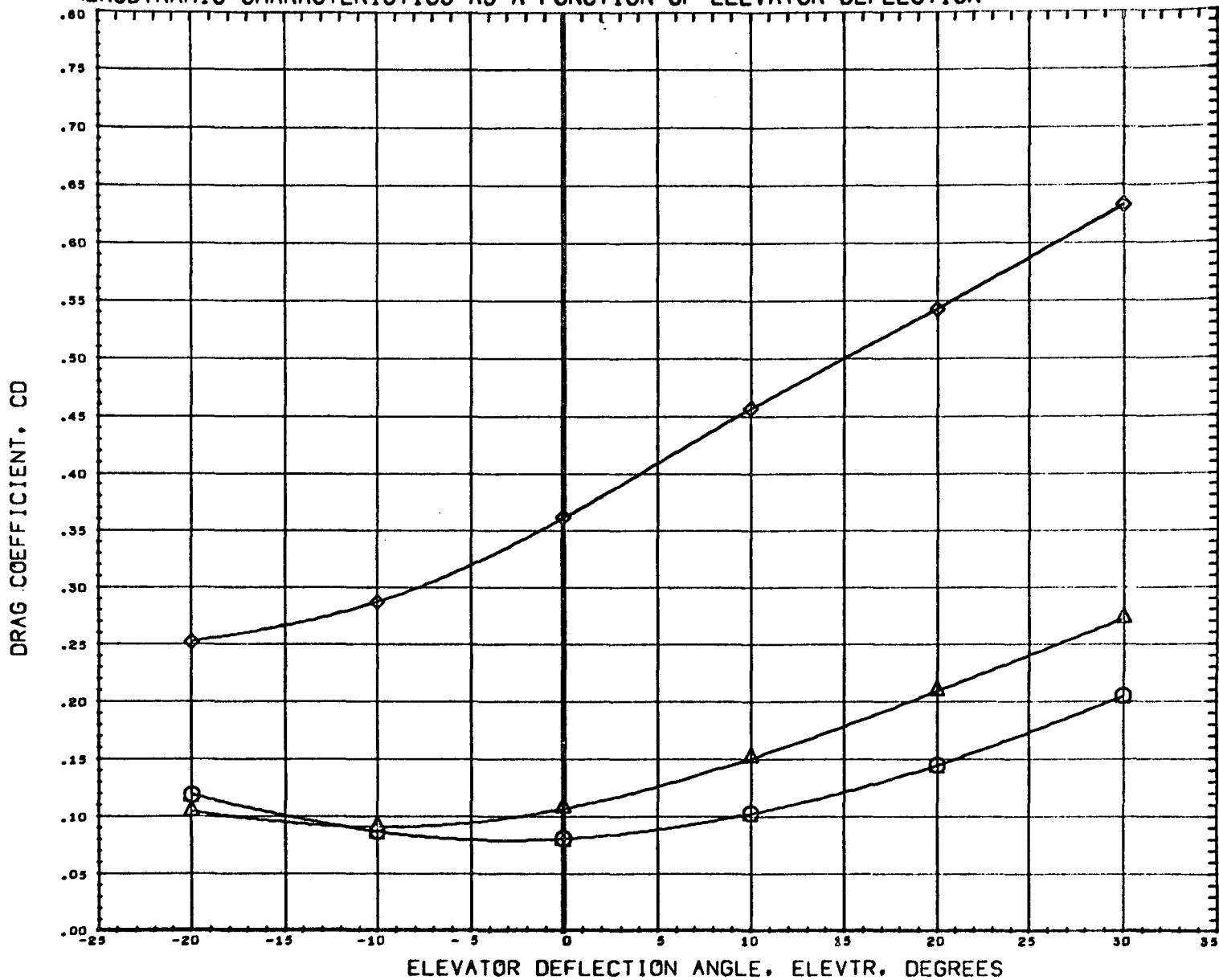
REFERENCE INFORMATION		
SREF	1.3550	50. FT.
LREF	3.4530	FT.
BREF	3.4530	FT.
XMRP	2.5950	FT.
YMRP	0.0000	FT.
ZMRP	0.0187	FT.
SCALE	1.9000	PER CT

DATA HIST. CODE MMI

CAL MSFC/LMSC BOOSTER B4C2F2W3V1

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AERODYNAMIC CHARACTERISTICS AS A FUNCTION OF ELEVATOR DEFLECTION



SYMBOL	ALPHA	PARAMETRIC VALUES			
	0.000	MACH	0.400	BETA	0.000
	6.000	CANARD	0.000	BSFLOW	0.000
	15.000				

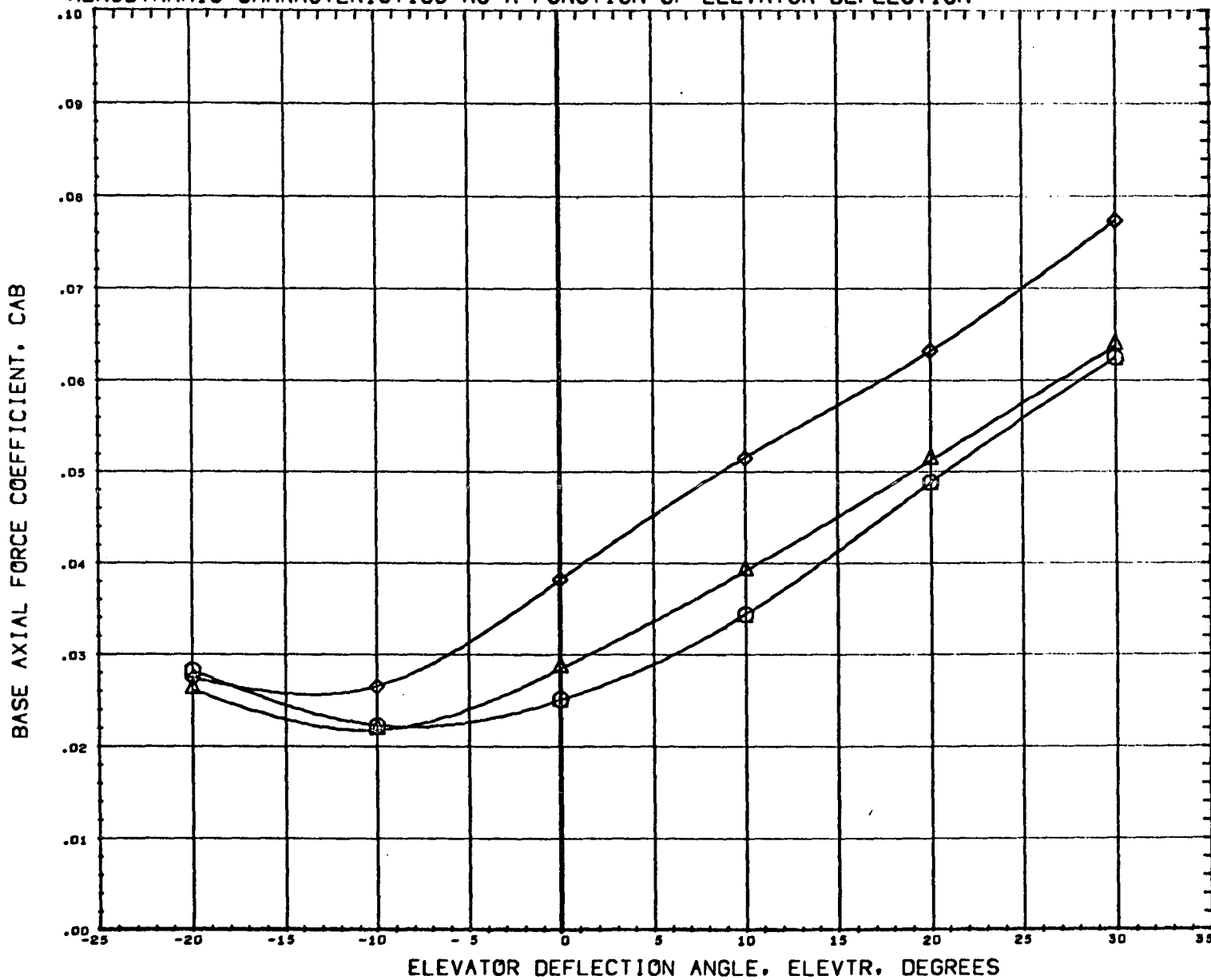
REFERENCE INFORMATION		
SREF	1.3550	SQ. FT.
LREF	3.4530	FT.
BREF	3.4530	FT.
XMRP	2.5950	FT.
YMRP	0.0000	FT.
ZMRP	0.0187	FT.
SCALE	1.3000	PER CT

DATA HIST. CODE MMI

CAL MSFC/LMSC BOOSTER B4C2F2W3V1

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AERODYNAMIC CHARACTERISTICS AS A FUNCTION OF ELEVATOR DEFLECTION



SYMBOL	ALPHA	MACH	PARAMETRIC VALUES		
○	0.000	0.400	BETA	0.000	
△	6.000	CANARD	0.000	BSFLOW	0.000
◇	15.000				

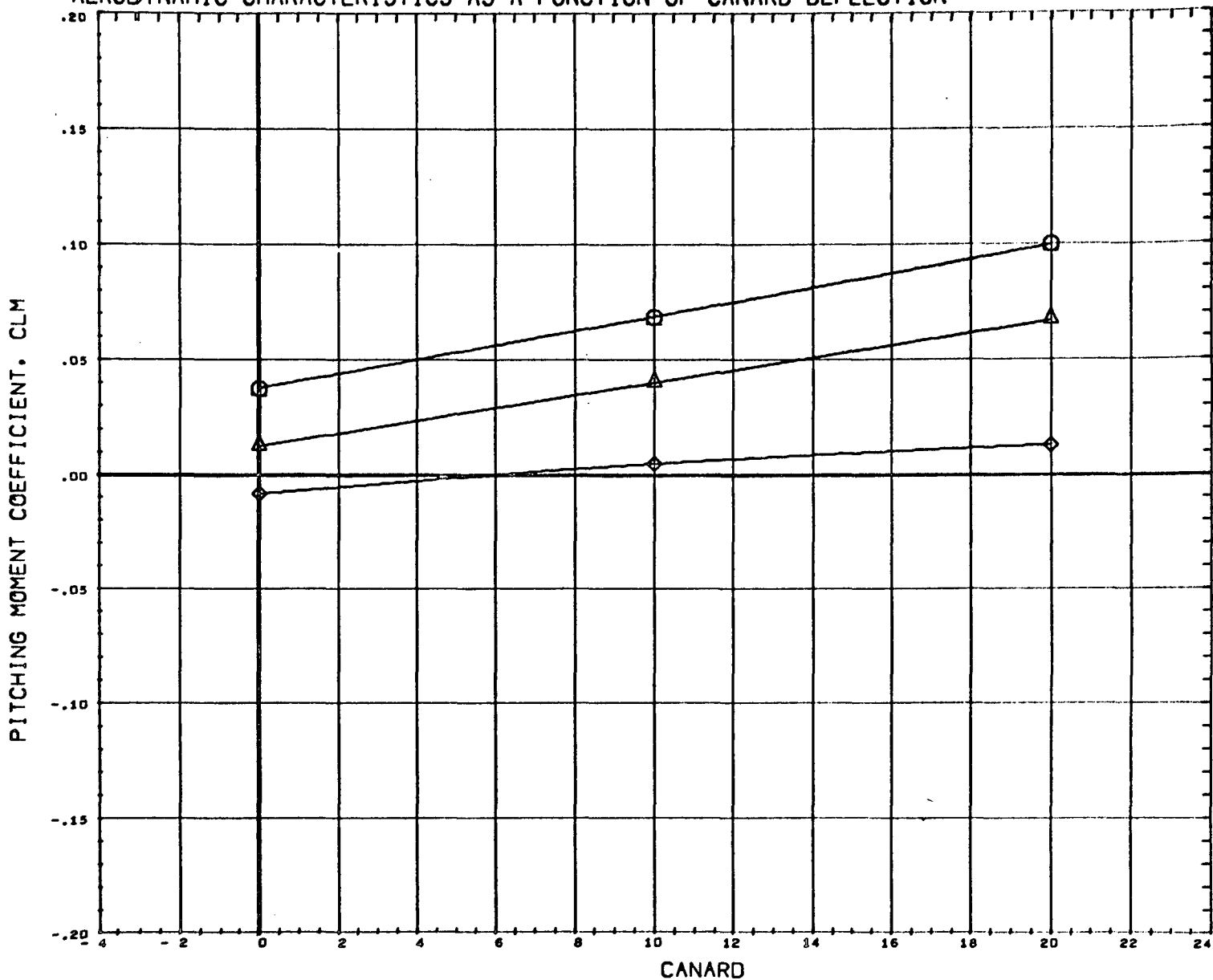
REFERENCE INFORMATION		
SREF	1.3550	89. FT.
LREF	3.4530	FT.
BREF	3.4530	FT.
XMRP	2.5950	FT.
YMRP	0.0000	FT.
ZMRP	0.0187	FT.
SCALE	1.5000	PER CT

DATA HIST. CODE MM1

CAL MSFC/LMSC BOOSTER B4C2F2W3V1

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AERODYNAMIC CHARACTERISTICS AS A FUNCTION OF CANARD DEFLECTION



SYMBOL	ALPHA	PARAMETRIC VALUES			
○	0.000	MACH	0.400	BETA	0.000
△	6.000	ELEVTR	0.000	BSFLOW	0.000
◇	15.000				

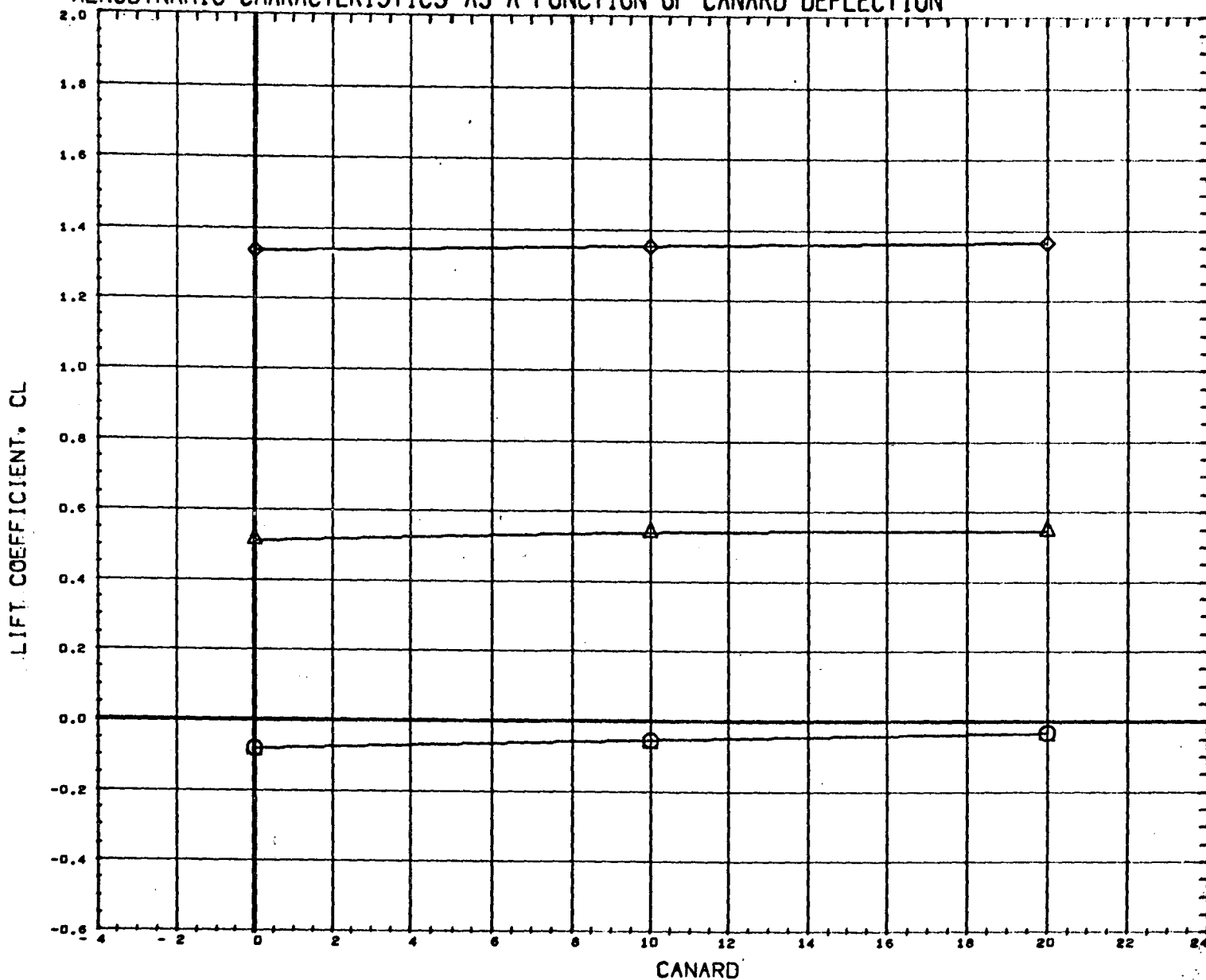
REFERENCE INFORMATION		
SREF	1.3550	SQ. FT.
LREF	3.4530	FT.
BREF	3.4530	FT.
XMRP	2.5950	FT.
YMRP	0.0000	FT.
ZMRP	0.0187	FT.
SCALE	1.5000	PER CT

DATA HIST. CODE MMI

CAL MSFC/LMSC BOOSTER B4C2F2W3V1

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AERODYNAMIC CHARACTERISTICS AS A FUNCTION OF CANARD DEFLECTION



SYMBOL	ALPHA	PARAMETRIC VALUES			
○	0.000	MACH	0.400	BETA	0.000
△	6.000	ELEVTR	0.000	BSFLOW	0.000
◇	15.000				

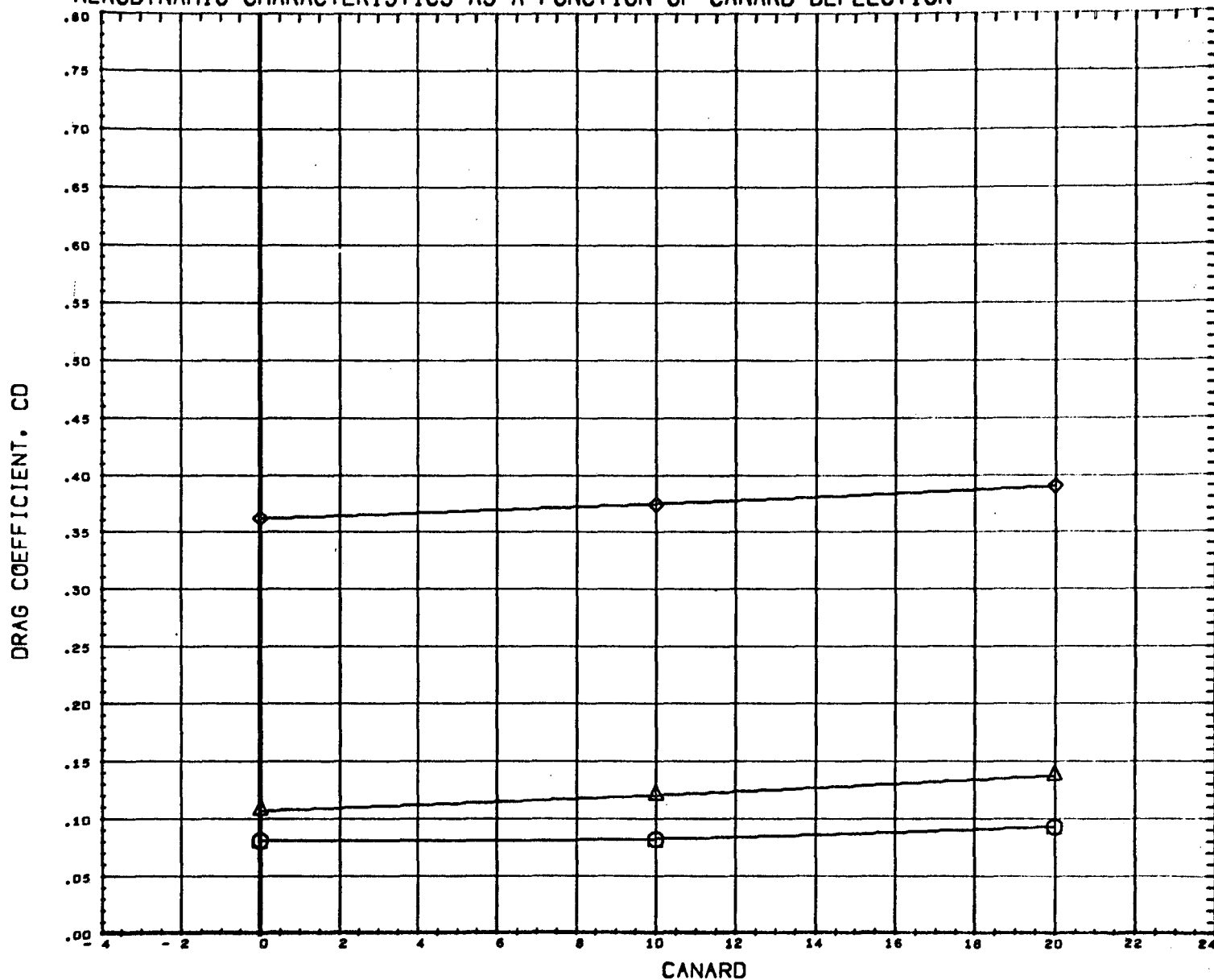
REFERENCE INFORMATION		
SREF	1.3550	89. FT.
LREF	3.4530	FT.
BREF	3.4530	FT.
XMRP	2.5950	FT.
YMRP	0.0000	FT.
ZMRP	0.0187	FT.
SCALE	1.5000	PER CT

DATA HIST. CODE HMI

CAL MSFC/LMSC BOOSTER B4C2F2W3V1

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AERODYNAMIC CHARACTERISTICS AS A FUNCTION OF CANARD DEFLECTION



SYMBOL	ALPHA	MACH	PARAMETRIC VALUES		
○	0.000	0.400	BETA	0.000	
△	6.000	ELEVTR	0.000	BSFLOW	0.000
◇	15.000				

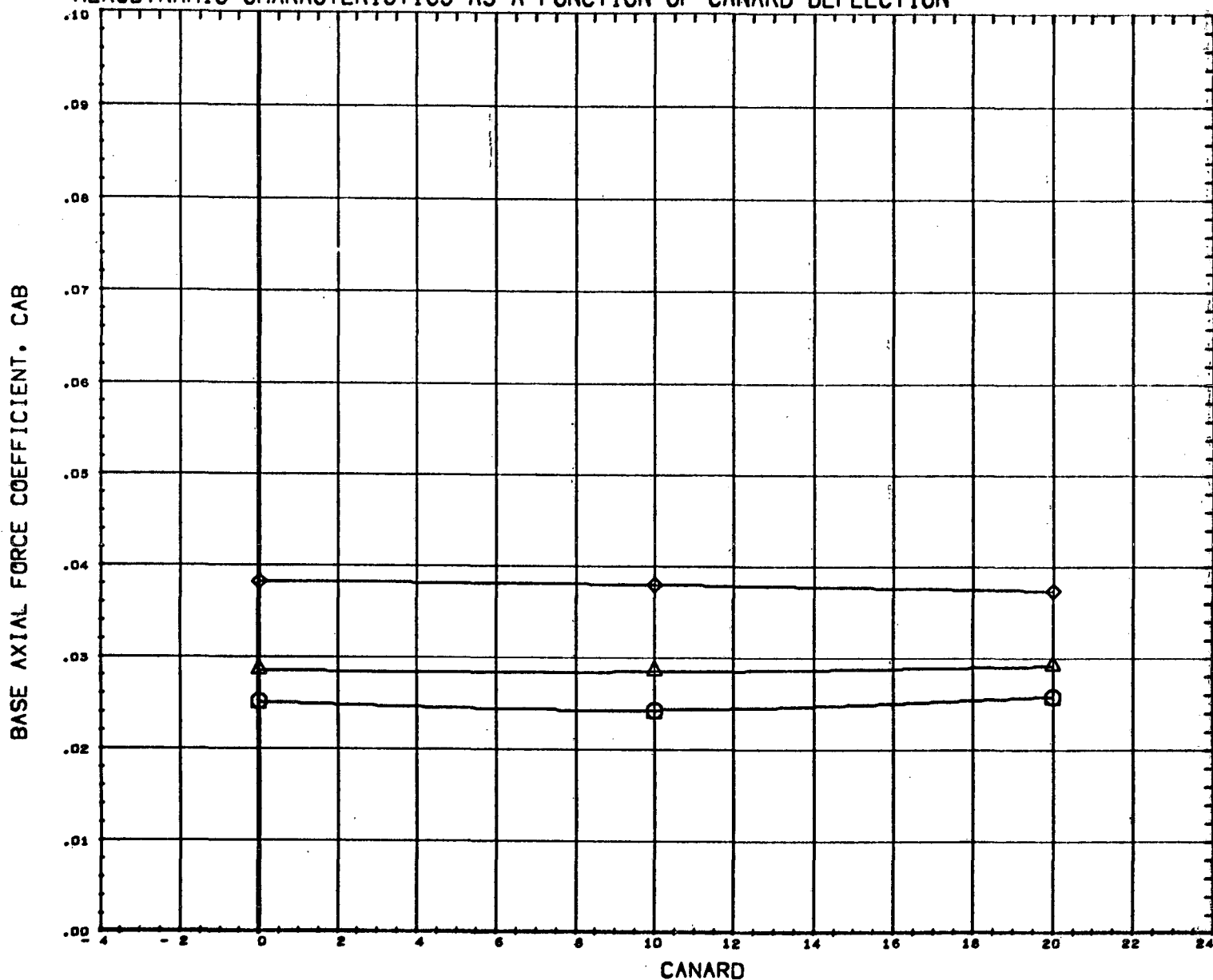
REFERENCE INFORMATION		
SREF	1.3550	SQ.FT.
LREF	3.4530	FT.
BREF	3.4530	FT.
XMRP	2.5950	FT.
YMRP	0.0000	FT.
ZMRP	0.0187	FT.
SCALE	1.5000	PER CT

DATA HIST. CODE MMI

CAL MSFC/LMSC BOOSTER B4C2F2W3V1

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AERODYNAMIC CHARACTERISTICS AS A FUNCTION OF CANARD DEFLECTION



SYMBOL	ALPHA	MACH	PARAMETRIC VALUES		
◇	0.000	0.400	BETA	0.000	
△	6.000	ELEVTR	0.000	BSFLOW	0.000
○	15.000				

REFERENCE INFORMATION		
SREF	1.3550	80. FT.
LREF	3.4530	FT.
BREF	3.4530	FT.
XMRP	2.5950	FT.
YMRP	0.0000	FT.
ZMRP	0.0167	FT.
SCALE	1.5000	PER CT

DATA HIST. CODE MMI

CAL MSFC/LMSC BOOSTER B4C2F2W3V1

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