

DMS-DR-1225
CR-120,046
VOLUME II
MARCH 1972

—SPACE SHUTTLE—



SURFACE PRESSURE AND
INVISCID FLOW FIELD PROPERTIES
OF THE NORTH AMERICAN ROCKWELL
DELTA-WING ORBITER FOR
NOMINAL MACH NUMBER OF 8

by

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(NASA-CR-120046) SURFACE PRESSURE AND
INVISCID FLOW FIELD PROPERTIES OF THE NORTH
AMERICAN ROCKWELL DELTA-WING ORBITER FOR
NOMINAL MACH NUMBER R.K. Matthews, et al
(Chrysler Corp.) Mar. 1972 59 p CSCL 22B G3/31

N72-22896

Unclassified
25161

VKF 50-INCH

HYPersonic TUNNEL B

Arnold Engineering
Development Center

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SADSAC SPACE SHUTTLE
AEROTHERMODYNAMIC
DATA MANAGEMENT SYSTEM

CONTRACT NAS8-4016
MARSHALL SPACE FLIGHT CENTER

SPACE DIVISION  CHRYSLER
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This document should
be referenced as
NASA CR-120-046

62 P80

CAT. 31

NASA Series Number: P-1007

DMS-DR- 1225
CR-120,046
VOLUME II
March, 1972

SADSAC/SPACE SHUTTLE

WIND TUNNEL TEST DATA REPORT

CONFIGURATION: North American Rockwell Delta Wing Orbiter

TEST PURPOSE: To Determine Surface Pressures and Inviscid Flow Field
Properties at Mach Number 8

TEST FACILITY: AEDC VKF 50-Inch Hypersonic Tunnel B

TESTING AGENCY: AEDC-MSFC

TEST NO. & DATE: VT 1162-7; September, 1971

FACILITY COORDINATOR: Mr. L. L. Trimmer, ARO, INC.

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CONTRACT NAS 8-4016

AMENDMENT 153

DRL 184 - 58

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FOREWORD

The work reported herein was sponsored by the Marshall Space Flight Center (MSFC), NASA. The results of tests presented were obtained by ARO, Inc. (a subsidiary of Sverdrup & Parcel and Associates, Inc.), contract operator of the Arnold Engineering Development Center (AEDC), AFSC, Arnold Air Force Station, Tennessee. Ascent and reentry conditions were simulated on shuttle models designed by McDonnell Douglas (MDAC), North American Rockwell (NAR) and General Dynamics Convair (GDC). In addition a limited amount of data were obtained on two research models provided by the Langley Research Center (LRC). Because of the broad scope of these tests the data will be presented in a series of SADSAC reports. This report presents the results of the surface pressure and flow field tests conducted at Mach 8 in Tunnel B on the North American Rockwell Delta Wing Orbiter.

CONTENTS

<u>Section</u>	<u>Page</u>
LIST OF FIGURES	iv
LIST OF TABLES	iv
NOMENCLATURE	v
1. INTRODUCTION	1
2. MODELS AND APPARATUS	1
2.1 Model Description	1
2.2 Facility Description	1
2.3 Instrumentation	2
3. PROCEDURE	2
3.1 Test Conditions	2
3.2 Data Reduction	2
3.3 Data Precision	3
4. DATA PRESENTATION	4

LIST OF FIGURES

<u>Figure</u>	<u>Page</u>
1. North American Rockwell Delta Wing Orbiter Model Sketch (0.013 Scale)	5
2. Model Photograph	6
3. Probe Rakes and Support	7

LIST OF TABLES

<u>Table</u>	
1. Configuration Description Details	8-13
2. Test Data Summary Sheets	14-15
3. Summary Data Plot Index	16

NOMENCLATURE

ALPHA-MODEL (α)	Model angle of attack, deg
ALPHA-PREBEND	Sting prebend angle, deg
ALPHA-SECTOR	Tunnel sector angle, deg
CP	Pressure coefficient, $(P_m - (P_\infty)) / Q_\infty$
CP-MAX	Pressure coefficient based on P_01 , $(P_01 - (P_\infty)) / Q_\infty$
L	Model length (29.25 in.)
MACH NO.	Free-stream Mach number
ML	Local Mach number
MU-INF	Free-stream viscosity, lb-sec/ft ²
MUL	Local viscosity, lb-sec/ft ²
P-INF	Free-stream pressure, psia
PM	Model surface pressure, psia
PML	Local model surface pressure, psia
P0	Tunnel stilling chamber pressure, psia
P01	Stagnation pressure downstream of a normal shock, psia
PR	Rake probe stagnation pressure, psia
Q-INF	Free-stream dynamic pressure, psia
RE/FT	Free-stream unit Reynolds number, ft ⁻¹
REL	Local unit Reynolds number, ft ⁻¹
RHO-INF	Free-stream density, lbm/ft ³
RHOL	Local density, lbm/ft ³
RHOUL	Local density-velocity product, lbm/ft ² -sec
ROLL-MODEL (ϕ)	Model roll angle, deg
T-INF	Free-stream temperature, °R

TL	Local temperature, °R
T0	Tunnel stilling chamber temperature, °R
TTR	Total temperature measured by rake probes, °R
U-INF	Free-stream velocity, ft/sec
UL	Local velocity, ft/sec
X	Axial coordinate (see Fig. 1), in.
Y	Distance from model surface or probe height (see Figs. 1 and 3), in.
YAW	Model yaw angle, deg

SECTION 1

INTRODUCTION

This report presents the results of a wind tunnel test program to determine surface pressures and flow field properties on the North American Rockwell orbiter configuration. The tests were conducted at the Arnold Engineering Development Center (AEDC) in Tunnel B of the von Karman Gas Dynamics Facility (VKF). The tests were conducted in September 1971.

Data were obtained at a nominal Mach number of 8 and a free-stream unit Reynolds number of 3.7×10^6 per foot. Angle of attack was varied from 10 to 50 deg in 10-deg increments.

SECTION 2

MODELS AND APPARATUS

2.1 MODEL DESCRIPTION

Model drawings were provided ARO, Inc. by the North American Rockwell Corporation and fabrication of the Stycast model was subcontracted to the Grumman Aircraft Corporation. The model had a 1.0-in. long steel nose and 10 windward centerline pressure orifices. A sketch showing the overall model dimensions is presented in Fig. 1 and a photograph of the configuration is shown in Fig. 2. Table 1 provides additional configuration description details but it should be pointed out that the models were cast as one smooth surface without moveable control surfaces.

2.2 FACILITY DESCRIPTION

Tunnel B is a continuous, closed-circuit, variable density wind tunnel with an axisymmetric contoured nozzle and a 50-in.-diam test section.

The tunnel can be operated at a nominal Mach number of 6 or 8 at stagnation pressures from 20 to 300 and 50 to 900 psia, respectively, and at stagnation temperatures up to 1350°R. The model may be injected into the tunnel for a test run and then retracted for model cooling or model changes without interrupting the tunnel flow.

2.3 INSTRUMENTATION

The model flow field was surveyed with pitot-pressure and single shield total temperature probe rakes. The rakes were mounted so that pressure and temperature measurements could be made simultaneously. The rakes, support mechanism, and spacing of the probes are shown in Fig. 3.

Static and pitot-probe pressures were measured with 15 psid transducers referenced to a near vacuum for pressures less than 15 psia and to atmospheric pressure for pressures greater than 15 psia. The atmospheric reference pressure was also measured with a 15 psid transducer.

SECTION 3

PROCEDURE

3.1 TEST CONDITIONS

Nominal test conditions are presented in the data summary sheets (Table 2) and the specific test conditions for each run are provided at the top of the data tabulation sheet for that run.

3.2 DATA REDUCTION

By assuming the flow-field static pressure equal to the wall static pressure, the local Mach number (M_L) was calculated from the Rayleigh pitot formula,

$$\frac{P_R}{P_{ML}} = \left(\frac{6ML^2}{5} \right)^{7/2} \left(\frac{6}{7ML^2 - 1} \right)^{5/2}, \text{ for } ML \geq 1$$

or from the compressible Bernoulli equation,

$$\frac{P_R}{P_{ML}} = (1 + 0.2 ML^2)^{7/2}, \text{ for } ML < 1.$$

The assumption of constant static pressure becomes less valid as the distance from the model surface increases.

The equations for the other flow field parameters are:

<u>Parameter</u>	<u>Equation</u>	<u>Units</u>
TL	$TL = \frac{T_0}{(1 + 0.2 ML^2)}$	°R
UL	$UL = (49.02)(ML) \sqrt{TL}$	ft/sec
RHOL	$RHOL = \frac{(2.70)(PML)}{TL}$	LBM/ft ³
MUL	$MUL = \frac{2.27 (TL)^{3/2}}{TL + 198.6} \times 10^{-8}$	lb-sec/ft ²
REL	$REL = \frac{(RHOL)(UL)}{(32.17)(MUL)}$	ft ⁻¹

The quantities calculated using TL are not valid in the model boundary layer since TTR is less than T₀ and, of course, none of the calculated parameters are meaningful outside the model shock layer.

3.3 DATA PRECISION

Estimated uncertainties of the primary measurements are given below:

<u>Parameter</u>	<u>Uncertainty</u>
PML	± 0.015 psia
P0	± 1.8 psia
P01	± 0.021 psia
PR	± 0.015 psia (for PR ≤ 15 psia) ± 0.021 psia (for PR > 15 psia)
T0	$\pm 10^\circ R$
TTR	$\pm 25^\circ R$

SECTION 4

DATA PRESENTATION

The test data are presented in tabulated and plotted form in Appendices A and B. The data are presented in the following order.

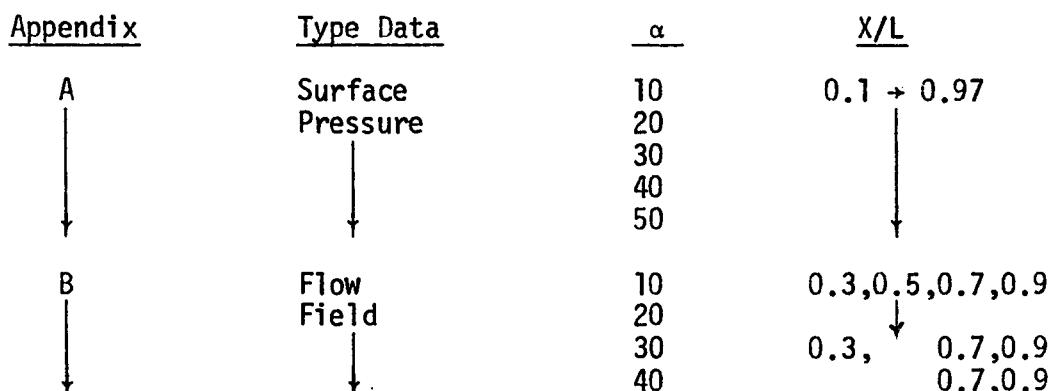


Table 3, Page 16, presents a summary of these data.

Pitot pressure and total temperature measurements were attempted at X/L stations of 0.7 and 0.9 at 50 degrees angle of attack; however, the rakes and support distorted the flow field as observed in shadowgraph photographs and therefore these measurements are not presented.

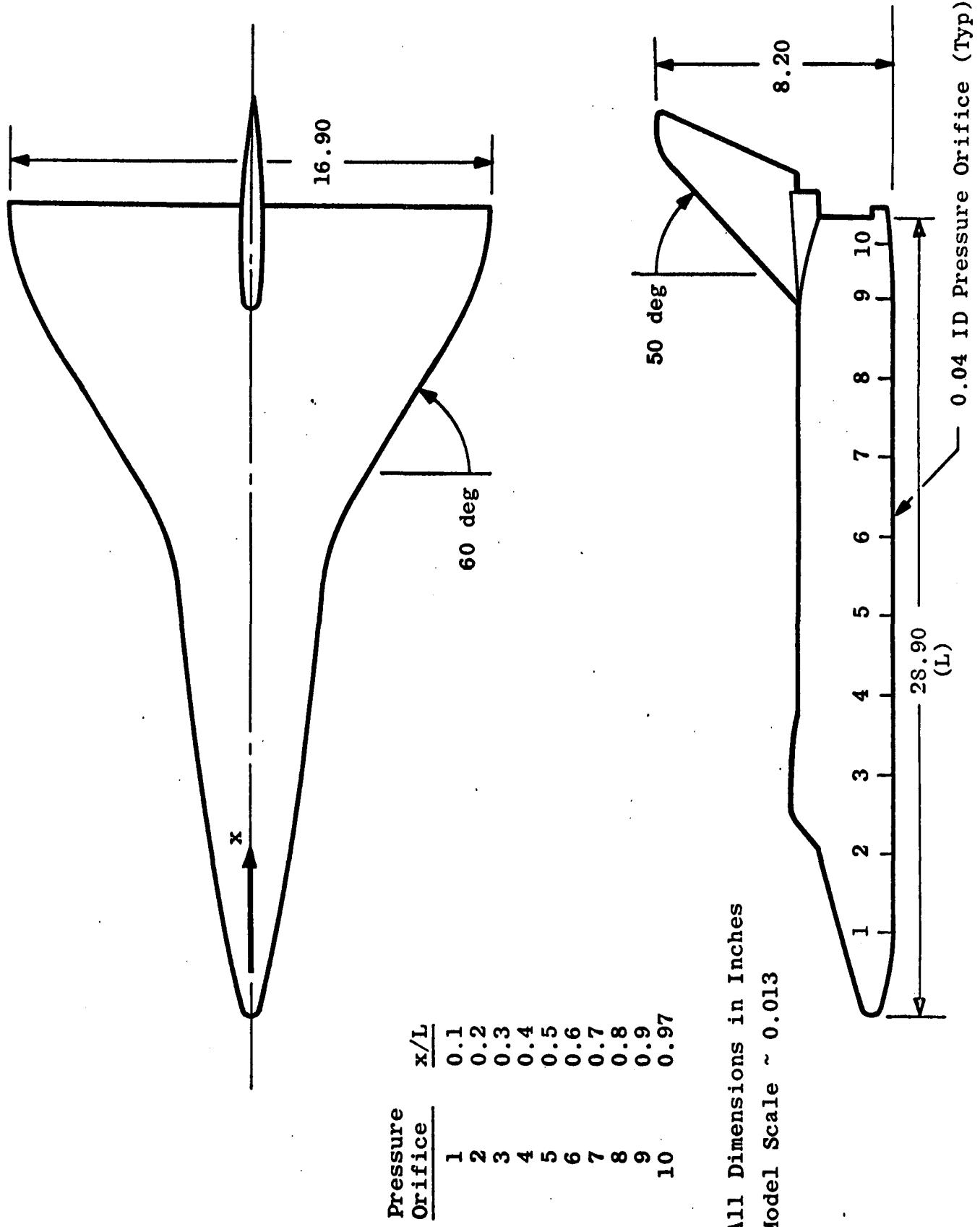
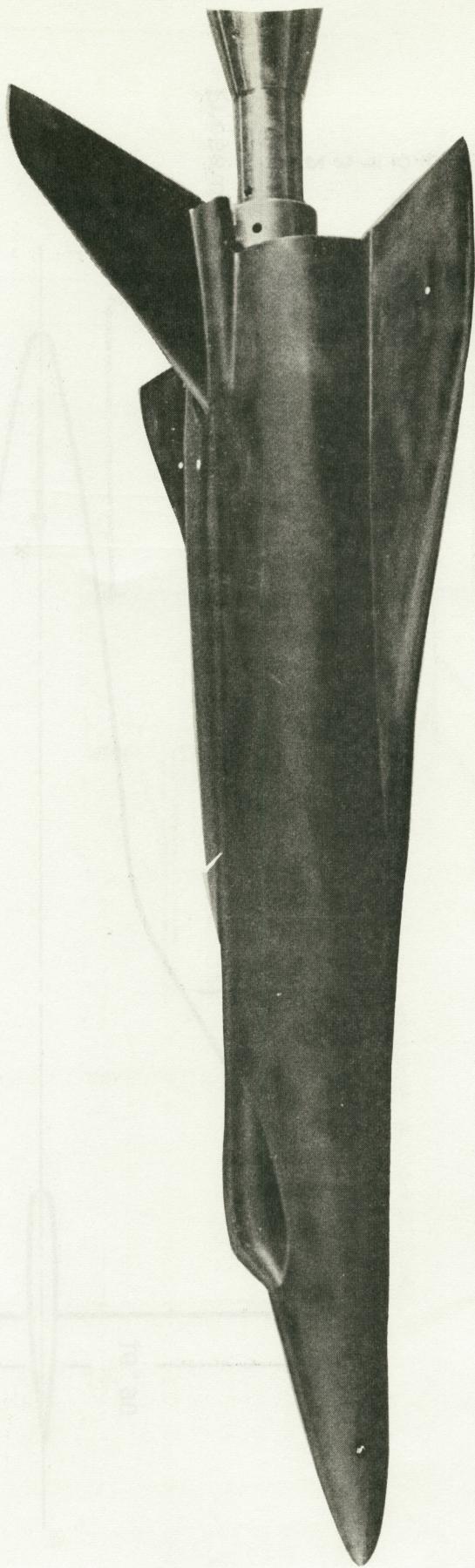


Fig. 1 North American Rockwell Delta Wing Orbiter Model Sketch (0.013 Scale)

Fig. 2 Model Photograph



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No.	Probe Height, Y, in.	Temperature Probes
1	0.014	0.051
2	0.066	0.131
3	0.112	0.202
4	0.163	0.303
5	0.216	0.402
6	0.258	0.599
7	0.313	
8	0.365	
9	0.415	
10	0.499	
11	0.606	
12	0.702	
13	0.802	
14	0.892	
15	0.981	

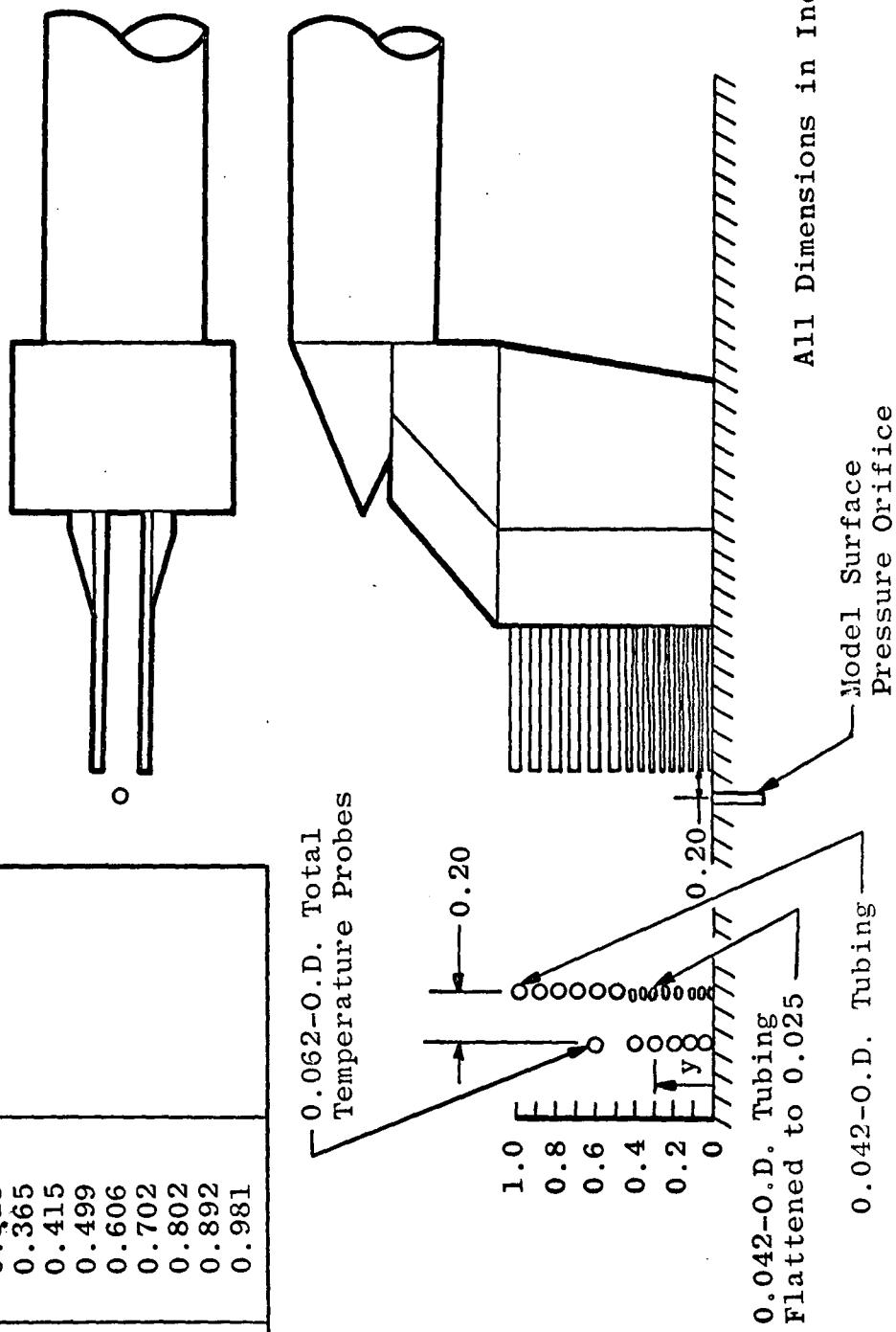


Fig. 3 Probe-Rakes and Support

TABLE 1
CONFIGURATION DESCRIPTION DETAILS

MODEL COMPONENT: BODY - B6

GENERAL DESCRIPTION: Basic delta wing fuselage. Fuselage reference plane is located at water plane 400.00 in.

Model scale = 0.013

DRAWING NUMBER: NR 9992-161B

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length	<u>2223.00</u>	<u>28.899</u>
Max. Width	<u>495.80</u>	<u>6.445</u>
Max. Depth	<u>263.00</u>	<u>3.419</u>
Fineness Ratio	<u>6.019</u>	<u>6.019</u>
Area		
Max. Cross-Sectional	<u>743.95</u>	<u>.1257</u>
Planform	<u>DNA</u>	<u>DNA</u>
Netted	<u>DNA</u>	<u>DNA</u>
Base	<u>DNA</u>	<u>DNA</u>

TABLE 1 - CONTINUED

MODEL COMPONENT: Wing - W₂₁

GENERAL DESCRIPTION: Delta wing with -5° twist and rounded wing tips. Wing blended into body. Used with body B6.

Model Scale = 0.013

DRAWING NUMBER: NR 9992-161B

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
--------------------	-------------------	--------------------

TOTAL DATA

Area, ft ²		
Planform	6511.00	1.100
Wetted	-	-
Span (equivalent), in.	1272.38	16.541
Aspect Ratio	1.714	1.714
Rate of Taper	1.719	1.719
Taper Ratio	0.144	0.144
Dihedral Angle, degrees	7.000	7.000
Incidence Angle, degrees	0.000	0.000
Aerodynamic Twist, degrees	-5.000	-5.000
Incidence, Root (B.P. 247.90)	0.000	0.000
Incidence, Tip (B.P. 557.70)	-5.000	-5.000
Sweep Back Angles, degrees		
Leading Edge	59.808	59.808
Trailing Edge	0.000	0.000
0.25 Element Line	52.197	52.197
Chords:		
Root (Wing Sta. 0.0)	1287.70	16.740
Tip, (equivalent) (W.S. 640.97)	186.00	2.418
MAC, inches (W.S. 240.62)	874.10	11.363
Fus. Sta. of .25 MAC	1793.32	23.313
W.P. of .25 MAC	280.73	3.649
Airfoil Section		
Root	NACA 0009-64	
Tip	NACA 0012-64	

EXPOSED DATA

Area, ft ²	3023.00	.5109
Span, (equivalent), in.	810.61	10.538
Aspect Ratio	1.498	1.498
Taper Ratio	0.209	0.209
Chords		
Root (Equiv.) (W.S. 232.62)	887.85	11.542
Tip (Equiv.) (W.S. 640.97)	186.00	2.418
MAC (W.S. 392.31)	613.38	7.974
Fus. Sta. of .25 MAC	1988.85	25.855
W.P. of .25 MAC	299.22	3.890

TABLE 1 - CONTINUED

MODEL COMPONENT: Vertical Tail - V27

GENERAL DESCRIPTION: Centerline vertical tail on delta wing configuration. The total data include the void area listed below. Used with body - B6.

Model Scale = 0.013

DRAWING NUMBER: NR 9992-161B

DIMENSIONS:FULL-SCALEMODEL SCALETOTAL DATA

Area, ft ²	626.03	.1058
Planform	1.99	.000336
*Void (included above)		
Span (equivalent), in.	361.06	4.694
Aspect Ratio	1.446	1.446
Rate of Taper	0.718	0.718
Taper Ratio	0.316	0.316
Dihedral Angle, degrees	-	-
Incidence Angle, degrees	-	-
Aerodynamic Twist, degrees	-	-
Toe-In Angle	0.000	0.000
Cant Angle	0.000	0.000
Sweep Back Angles, degrees		
Leading Edge	50.003	50.003
Trailing Edge	25.352	25.352
0.25 Element Line	45.352	45.352
Chords:		
Root (W.P. 511.62)	379.31	4.931
Tip, (equivalent)(W.P. 872.67)	120.05	1.561
MAC (W.P. 660.90)	272.11	3.537
Fus. Sta. of .25 MAC	2422.61	
W.P. of .25 MAC	660.90	8.592
Airfoil Section		
Root (W.P. 500.44)	NACA 0012-64	
Tip (W.P. 878.00)	NACA 0009-64	

EXPOSED DATA

Area	_____	_____
Span, (equivalent)	_____	_____
Aspect Ratio	_____	_____
Taper Ratio	_____	_____
Chords	_____	_____
Root	_____	_____
Tip	_____	_____
MAC	_____	_____
Fus. Sta. of .25 MAC	_____	_____
W.P. of .25 MAC	_____	_____

*This area is the void area located at the lower aft portion of the surface.

TABLE 1 - CONTINUED

MODEL COMPONENT: Elevon - E₁₁ (Data for one of two sides)GENERAL DESCRIPTION: Constant chord elevon located on delta wing - W₂₁Model Scale = 0.013DRAWING NUMBER: NR 9992-161B

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area (true), ft ²	423.09	0.0715
Span (equivalent), in.	417.30	5.425
Inb'd equivalent chord (W.S. 237.48)	146.00	1.898
Outb'd equivalent chord (W.S. 654.78)	146.00	1.898
Ratio Moveable Surface Chord/ Total Surface Chord		
At Inb'd equiv. chord	0.166	0.166
At Outb'd equiv. chord	0.900	0.900
Sweep Back Angles, degrees		
Leading Edge	0.000	0.000
Tailing Edge	0.000	0.000
Hingeline	0.000	0.000
Area Moment (Normal to hinge line), ft ³ (Product of area and mean chord)	5144.00	0.01130

TABLE 1 - CONTINUED

MODEL COMPONENT: Orbital Maneuvering System Shroud - ZGENERAL DESCRIPTION: Fairing over orbital maneuvering system. Located on aft upper fuselage mold line.Model Scale = 0.013DRAWING NUMBER: NR 9992-161B

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length (along upper surface), in.	<u>359.31</u>	<u>4.671</u>
Sta. of Leading Edge, in.	<u>2163.33</u>	
Sta. of Trailing Edge, in.	<u>2523.56</u>	
Pitch Angle (T.E. Up), deg.	<u>3.181</u>	<u>3.181</u>
Area		
Max. Cross-Sectional		
Planform		
Wetted		
Base		

TABLE 1 - CONCLUDED

MODEL COMPONENT: Drag Brake - J4 (Data for one of two sides)

GENERAL DESCRIPTION: Drag Brake - J4 is the deflectable side panel of the vertical tail V27 hinged at 60% element line and extending to the trailing edge.

Model Scale = 0.013

DRAWING NUMBER: NR 9992-161B

(All dimensions are in the drag brake reference plane)

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area, ft ²	242.39	.0410
Span (equivalent), in.	355.61	4.623
Inb'd equivalent chord , in. (W.P. 520.18)	149.22	1.940
Outb'd equivalent chord , in.	47.08	.612
Ratio Elevator chord/vertical tail chord		
At Inb'd equiv. chord	-	-
At Outb'd equiv. chord	-	-
Sweep Back Angles, degrees		
Leading Edge	37.273	37.273
Tailing Edge	25.352	25.352
Hingeline	37.273	37.273
Area Moment (Normal to hinge line), ft ³ (Product of area and mean chord)	1921.27	0.00422

TABLE 2.
TEST DATA SUMMARY SHEETS

TEST TITLE: NAR-DW0 Flow Field Tests

TEST NUMBER: VT1162

TEST FACILITY: AEDC Tunnel B

TEST DATE: September 1971

TEST ENGINEER: R. K. Matthews & W. R. Martindale

Run No.	Model Configuration Identification	Model Scale	Free Stream Mach Number	Total Pressure (psia)	Total Temp. ($^{\circ}$ R)	$Re/ft \times 10^{-6}$	Flow Field Survey Station X/L	Type Data*	Model Position (degrees)	Remarks.
320	NAR-DW0	0.013	8.0	860	1340	3.7	N/A	SP	10 0	180
321										
322										
323										
324										
316								0.3 FF	10	
312								0.5		
308								0.7		
303								0.9		
317								0.3	20	
315								0.5		
309								0.7		
306								0.9		

*SP - Surface Pressure
FF - Flow Field

TABLE 2. CONCLUDING

TEST DATA SUMMARY SHEETS

NAB-DWG Field Tests

VT1162

Contents

TEST DATE: _____

TEST FACILITY: AEDC Tunnel B

September 1971 TEST ENGINEER: R. K. Matthews & W. R. Martindale

Run No.	Model Configuration Identification	Model Scale	Free Stream Mach Number	Total Pressure (psia)	Total Temp. ($^{\circ}$ R)	$Re/ft_x \times 10^{-6}$	Flow Field Survey Station: X/L	Type Data	Model Position (degrees)	Remarks:
								α	ϕ	
318	NAR-DMO	0.013	8.0	860	1340	3.7	0.3	FF	30	0 180
311								0.7		
307								0.9		
325								0.7	40	
327								0.9		

*SP = Surface Pressure
FF = Flow Field

TABLE 3
SUMMARY DATA PLOT INDEX

TYPE OF DATA	PAGES	ANGLE OF ATTACK - DEGREES				FLOW FIELD SURVEY STATION (X/L)			
		10	20	30	40	50	0.3	0.5	0.7
SURFACE PRESSURE	19 21 23 25 27	X	X	X	X	X	N/A		
FLOW FIELD	30 32 34 36	X	X	X	X	X	X	X	X
	38 40 42 44	X	X	X	X	X	X	X	X
	46 48 50 52 54	X	X	X	X	X	X	X	X

SURFACE PRESSURE

PM/POL vs. X/L

Y vs. PR/POL
Y vs. TTR/T0
Y vs. ML

FLOW FIELD

Y vs. UL/UT-TMP
Y vs. RHOUL/RHOU

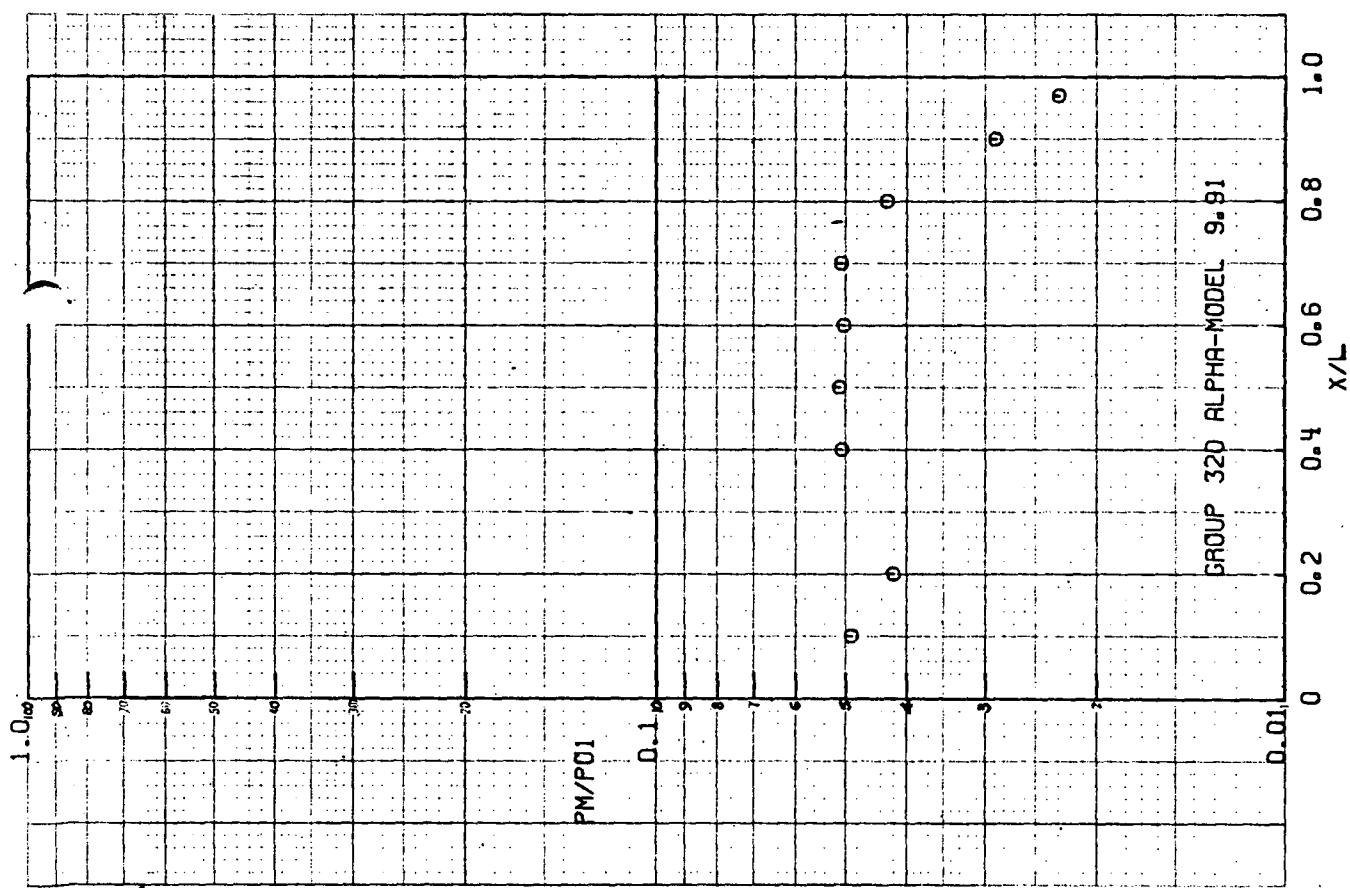
APPENDIX A
SURFACE PRESSURE

PAGE # 1

10/21/71

AEDC (ARO, INC.) ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL A
 VT1162

GROUP	CONFIG	MODEL	MACH NO.	P0 PSIA	10 DEG R ALPHA-MODEL	ALPHA-SECTOR	ROLL-MODEL	YAW
					9.91	13.09	-23.00	180.00
320	53	NAR-DW0	8.00	862.8	1342	9.91	13.09	0.0
T-INF	P-INF	P01	0-INF					
(DEG R)	(PSIA)	(PSIA)	(FT/SEC)					
97	8.84E-02	7.323	3.959					
Cr	POS	TAP	PM	PM/PO	PM/PO1	PM/P-INF	CP	CP/CP-MAX
			(PSIA)					X/L
1	2	1	3.585E-01	4.155E-04	4.896E-02	4.057E-00	6.823E-02	.100
2	2	2	3.077E-01	3.566E-04	4.201E-02	3.481E-00	5.539E-02	.031E-02
3	4	3.710E-01	4.300E-04	5.066E-02	4.198E-00	7.133E-02	3.907E-02	.200
4	5	3.750E-01	4.347E-04	5.221E-02	4.243E-00	7.244E-02	3.962E-02	.400
5	2	6	3.688E-01	4.274E-04	5.036E-02	4.173E-00	7.083E-02	.500
6	2	7	3.719E-01	4.311E-04	5.079E-02	4.209E-00	7.162E-02	.600
7	2	8	3.151E-01	3.652E-04	4.302E-02	3.565E-00	5.726E-02	.700
8	2	9	2.115E-01	2.451E-04	2.888E-02	2.393E-00	3.110E-02	.800
9	2	10	1.681E-01	1.948E-04	2.295E-02	1.902E-00	2.013E-02	.900
10								.970

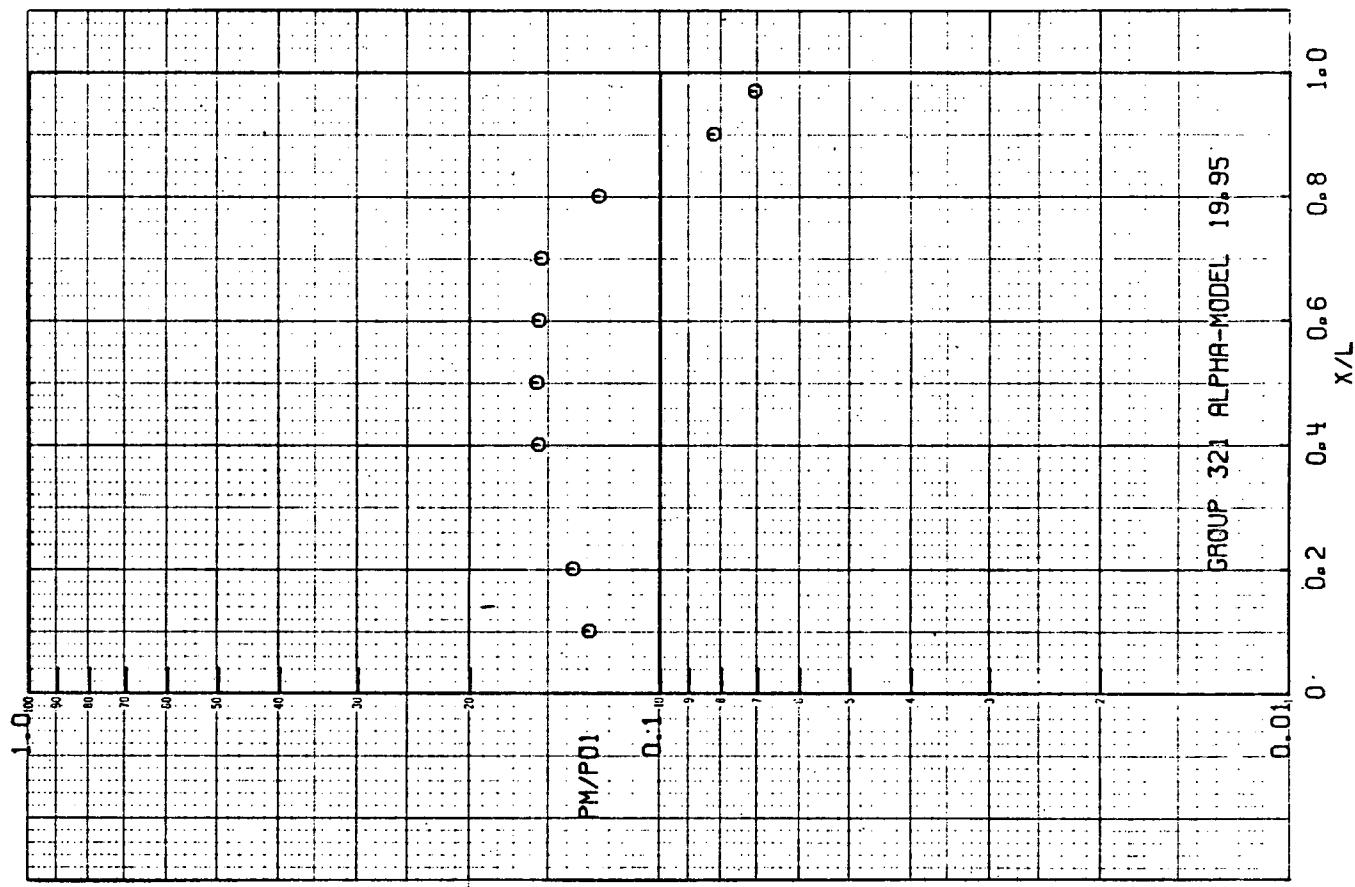


PAGE = 1

10/21/77

AEDC (ARO, INC.) ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL B
 VT1162

GROUP	CONFIG	MODEL	MACH NO.	P0 PSIA	T0 DEG R	ALPHA-MODEL	ALPHA-SECTOR	ROLL-MODEL	YAW
		NAR-DNO	8.00	862.3	1342	19.96	3.04	-23.00	180.00
321	53	T=INF (DEG R)	P01 (PSIA) 97	0-INF (PSIA) 8.83E-02	U-INF (FT/SEC) 7.319	RHO-INF (LBN/FT3) 3.957	MU-INF (LBN/FT2) 2.451E-03	RE/FT (FT-1) 7.829E-08	L (IN) 3.765E 06 28.90
		Cx	POS	TAP	PM	PM/PO	PM/P01	CP/PM-INF	CP
				(PSIA)					CP/CP-MAX
									X/L
1	2	1	9.480E-01	1.099E-03	1.295E-01	1.073E 01	2.173E-01	1.189E-01	.100
2	2	2	1.007E 00	1.0168E-03	1.376E-01	1.140E 01	2.322E-01	1.271E-01	.200
4	2	4	1.141E 00	1.023E-03	1.559E-01	1.292E 01	2.660E-01	1.456E-01	.400
5	2	5	1.149E 00	1.033E-03	1.570E-01	1.301E 01	2.681E-01	1.467E-01	.500
6	2	6	1.139E 00	1.021E-03	1.557E-01	1.290E 01	2.656E-01	1.424E-01	.600
7	2	7	1.129E 00	1.031E-03	1.543E-01	1.279E 01	2.631E-01	1.444E-01	.700
8	2	8	9.167E-01	1.063E-03	1.253E-01	1.038E 01	2.094E-01	1.144E-01	.800
9	2	9	6.030E-01	6.093E-04	8.239E-02	6.827E 00	1.301E-01	7.118E-02	.900
10	2	10	5.177E-01	6.004E-04	7.074E-02	5.9862E 00	1.085E-01	5.9395E-02	.970

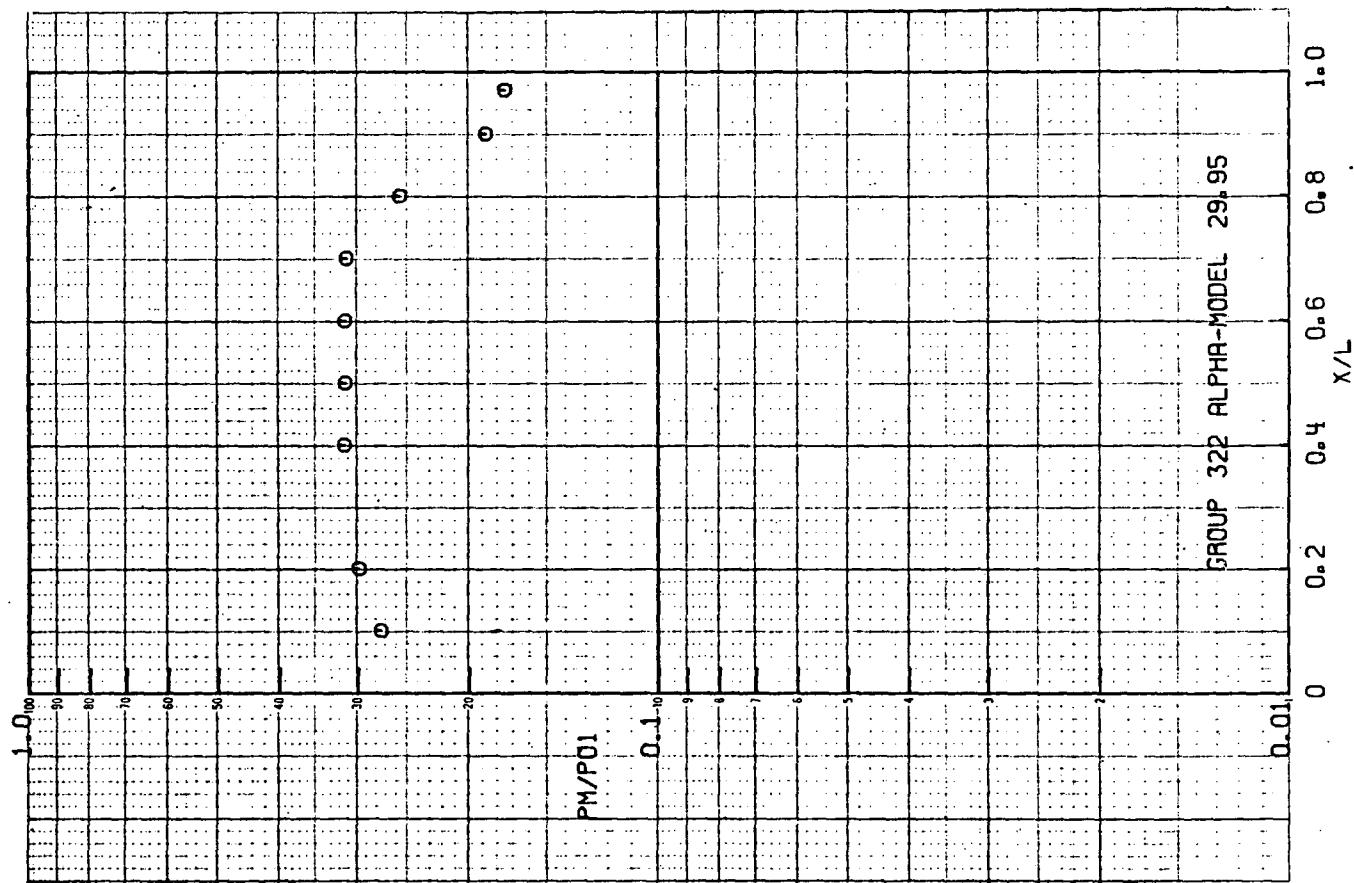


PAGE # 1

10/21/71

AEDC (ARO, INC.) ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL B
 VT1162

GROUP	CONFIG	MODEL	MACH NO.	P0	PSIA	TO DEG R	ALPHA-MODEL	ALPHA-SECTOR	ROLL-MODEL	YAW
322	53	NAR-DWO	8.00	861.7	1342	29.95	6.95	23.00	18.00	0
T-1NF	P-1NF	P01		0-INF	U-INF	RHO-INF	MU-INF	RE/FT	L	
(DEG R)	(PSIA)	(PSIA)		(FT/SEC)	(LB/SEC/FT2)	(LB/SEC/FT3)	(IN)	(FT-1)		
97	8.83E-02	7.314	3.954	3867	2.450E-03	7.829E-08	3.762E-06	28.90		
Cr	POS	TAP	PM	PM/PO	PM/PO1	PM/P-1NF	CP	CP/CP-MAX	X/L	
			(PSIA)							
1	2	1	2.013E 00	2.336E-03	2.752E-01	2.281E 01	4.868E-01	2.664E-01	.100	
2	2	2	2.182E 00	2.532E-03	2.084E-01	2.472E 01	5.290E-01	2.898E-01	.200	
3	4	2	2.298E 00	2.667E-03	3.142E-01	2.604E 01	5.589E-01	3.058E-01	.400	
4	2	4	2.292E 00	2.660E-03	3.134E-01	2.597E 01	5.571E-01	3.050E-01	.500	
5	2	5	2.294E 00	2.662E-03	3.136E-01	2.599E 01	5.578E-01	3.052E-01	.600	
6	2	6	2.280E 00	2.6446E-03	3.117E-01	2.583E 01	5.542E-01	3.033E-01	.700	
7	2	7	2.180E 00	2.596E-03	2.568E-01	2.128E 01	4.527E-01	2.478E-01	.800	
8	2	8	1.878E 00	2.180E-03	1.881E-01	1.558E-01	3.254E-01	1.782E-01	.900	
9	2	9	1.375E 00	1.596E-03	1.469E-01	1.754E-01	1.454E-01	1.654E-01	.970	
10	2	10	1.283E 00	1.469E-03						

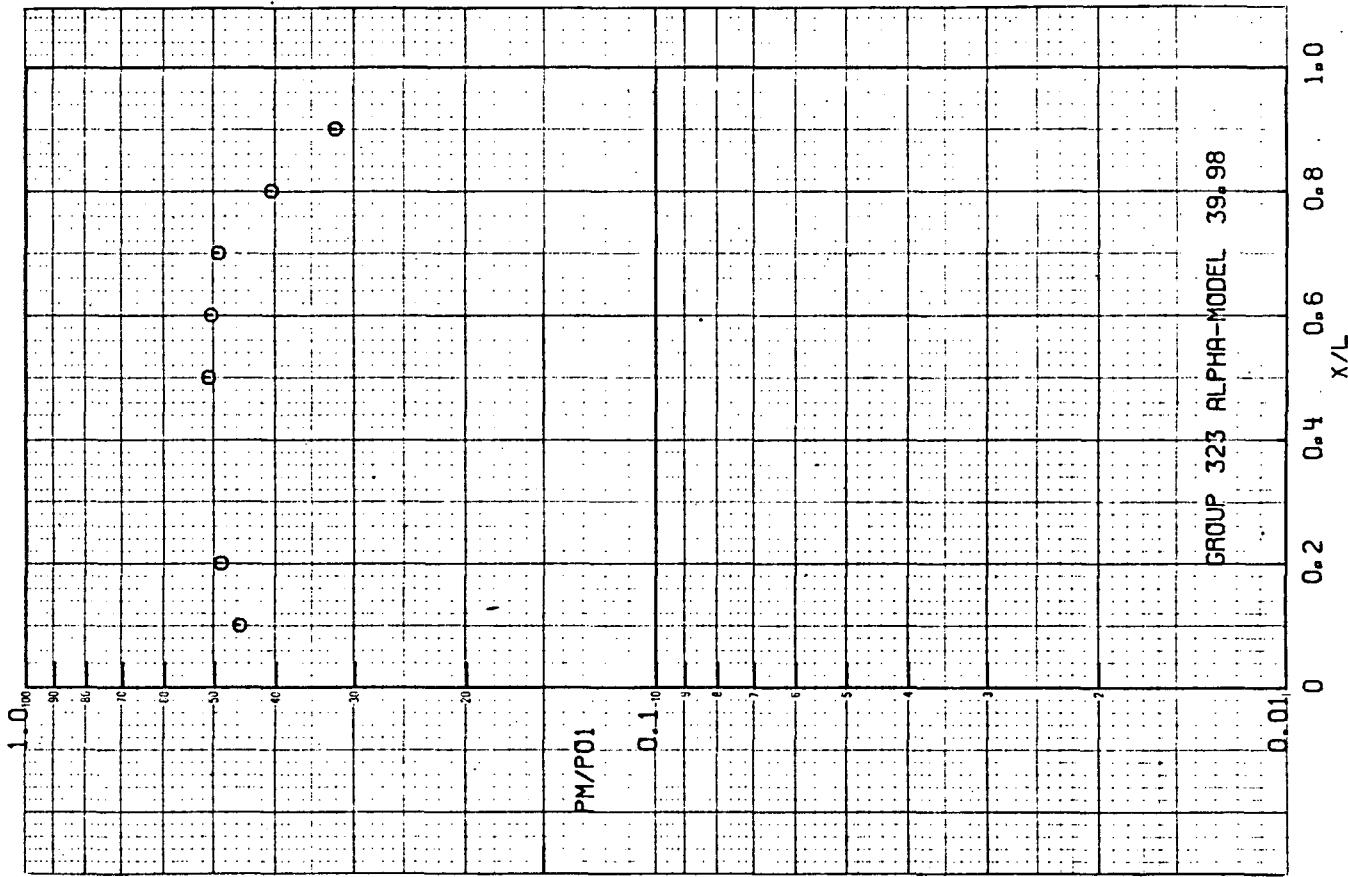


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10/21/71

AEDC TARO, INC. ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL A
 VT1162

GROUP	CONFIG	MODEL	MACH NO.	P0 PSIA	10 DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW
		NAR-DW0	8.00	861.4	1341	39.98	10.02	250.00	180.00	0
T=INF	P=INF	P01	0-INF	U-INF	RHO-INF	MU-INF	RE/FT	L		
(DEG R)	(PSIA)	(PSIA)	(FT/SEC)	(FT/SEC)	(LB-SEC/FT2)	(LEM/FT3)	(FT-1)	(IN)		
97	0.82E-02	7.311	3.953	3865	2.451E-03	7.823E-08	3.765E-06	28.90		
CH	POS	TAP	PM	PM/P0	PM/PM-1NF	CP	CP/CP-MAX	X/L		
		(PSIA)								
1	2	1	3.334E-00	3.871E-03	4.561E-01	3.779E-01	8.213E-01	4.494E-01	.100	
	2	2	3.570E-00	4.145E-03	4.883E-01	4.047E-01	8.810E-01	4.882E-01	.200	
	5	2	3.733E-00	4.334E-03	5.107E-01	4.232E-01	9.222E-01	5.047E-01	.500	
	6	2	3.692E-00	4.287E-03	5.050E-01	4.185E-01	9.118E-01	4.890E-01	.600	
	7	2	3.598E-00	4.177E-03	4.921E-01	4.078E-01	8.879E-01	4.859E-01	.700	
	8	2	2.968E-00	3.445E-03	4.059E-01	3.364E-01	7.285E-01	3.986E-01	.800	
	9	2	2.347E-00	2.725E-03	3.211E-01	2.661E-01	5.716E-01	3.128E-01	.900	

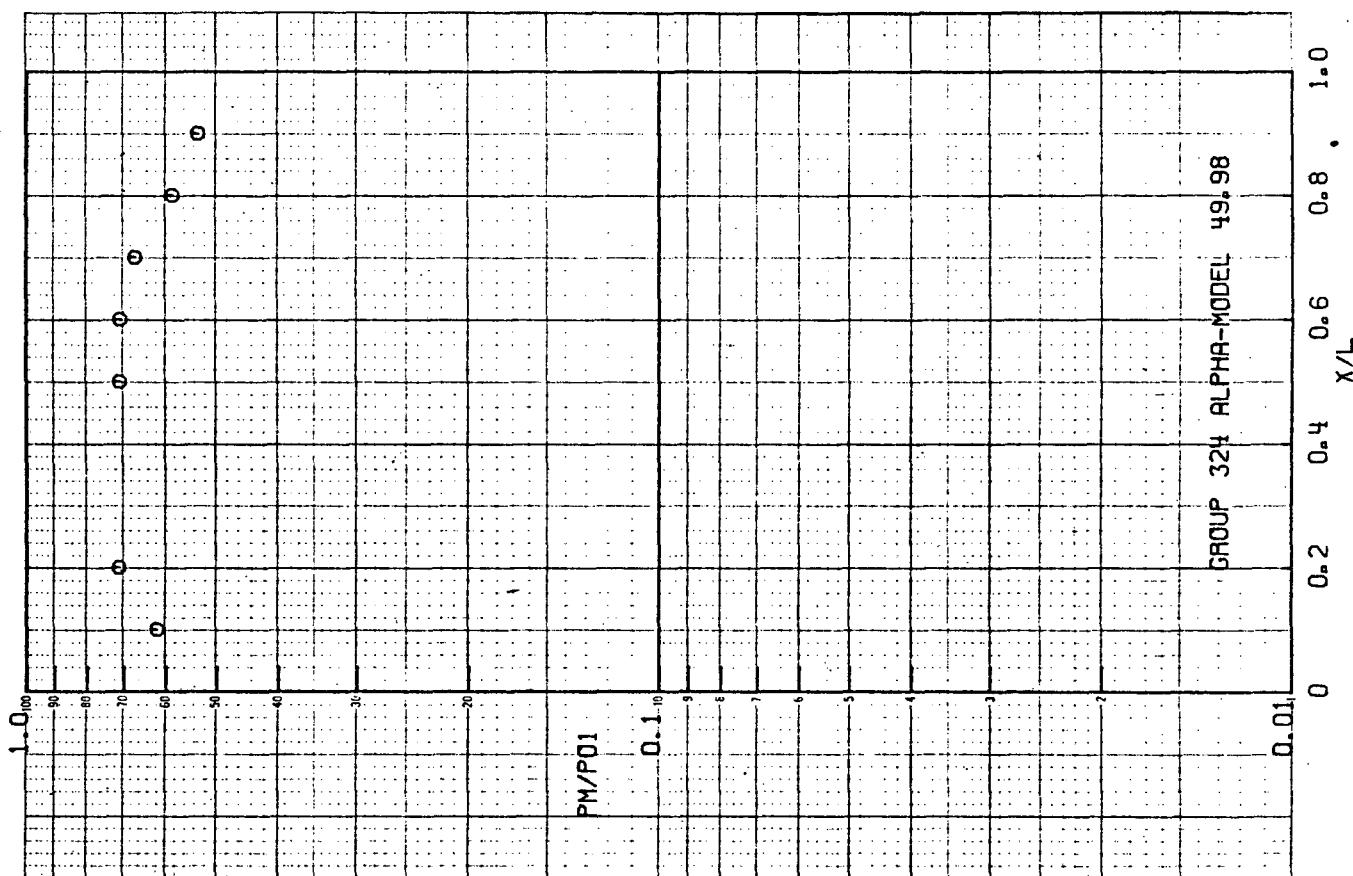


PAGE # 1

10/21/71

AEDC (ARO) INC. ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL A
 VTL162

GROUP	CONFIG	MODEL	MACH NO.	P0 PSIA	TO DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW
324	53	NAR-DW0	8.00	859.8	1340	49.99	.01	-50.00	180.00	.0
T- INF	P- INF	P01	0-INF	U-INF	RHO-INF	MU-INF	RE/FT	L		
(DEG R)	(PSIA)	(PSIA)	(PSIA)	(FT/SEC)	(LBX/FT3)	(IN)	(FT-1)	(IN)		
97	8.91E-02	7.297	3.945	3864	2.448E-03	7.818E-08	3.762E-06	28.90		
CH	POS	TAP	PM	PM/PO	PM/PO1	PM/P-INF	CP	CP/CP-MAX	X/L	
			(PSIA)							
1	2	1	4.529E 00	5.267E-03	6.206E-01	5.142E 01	1.126E 00	6.159E-01	.100	
2	2	2	5.211E 00	6.061E-03	7.140E-01	5.917E 01	1.294E 00	7.105E-01	.200	
3	2	5	5.88E 00	6.035E-03	7.110E-01	5.892E 01	1.293E 00	7.055E-01	.500	
4	2	6	5.191E 00	6.038E-03	7.114E-01	5.895E 01	1.294E 00	7.059E-01	.600	
5	2	7	4.918E 00	5.720E-03	6.739E-01	5.584E 01	1.224E 00	6.699E-01	.700	
6	2	8	4.289E 00	4.999E-03	5.878E-01	4.871E 01	1.065E 00	5.848E-01	.800	
7	2	9	3.910E 00	4.548E-03	5.359E-01	4.451E 01	9.689E-01	5.302E-01	.900	



APPENDIX B

FLOW FIELD DATA

Page 1

00/20/71

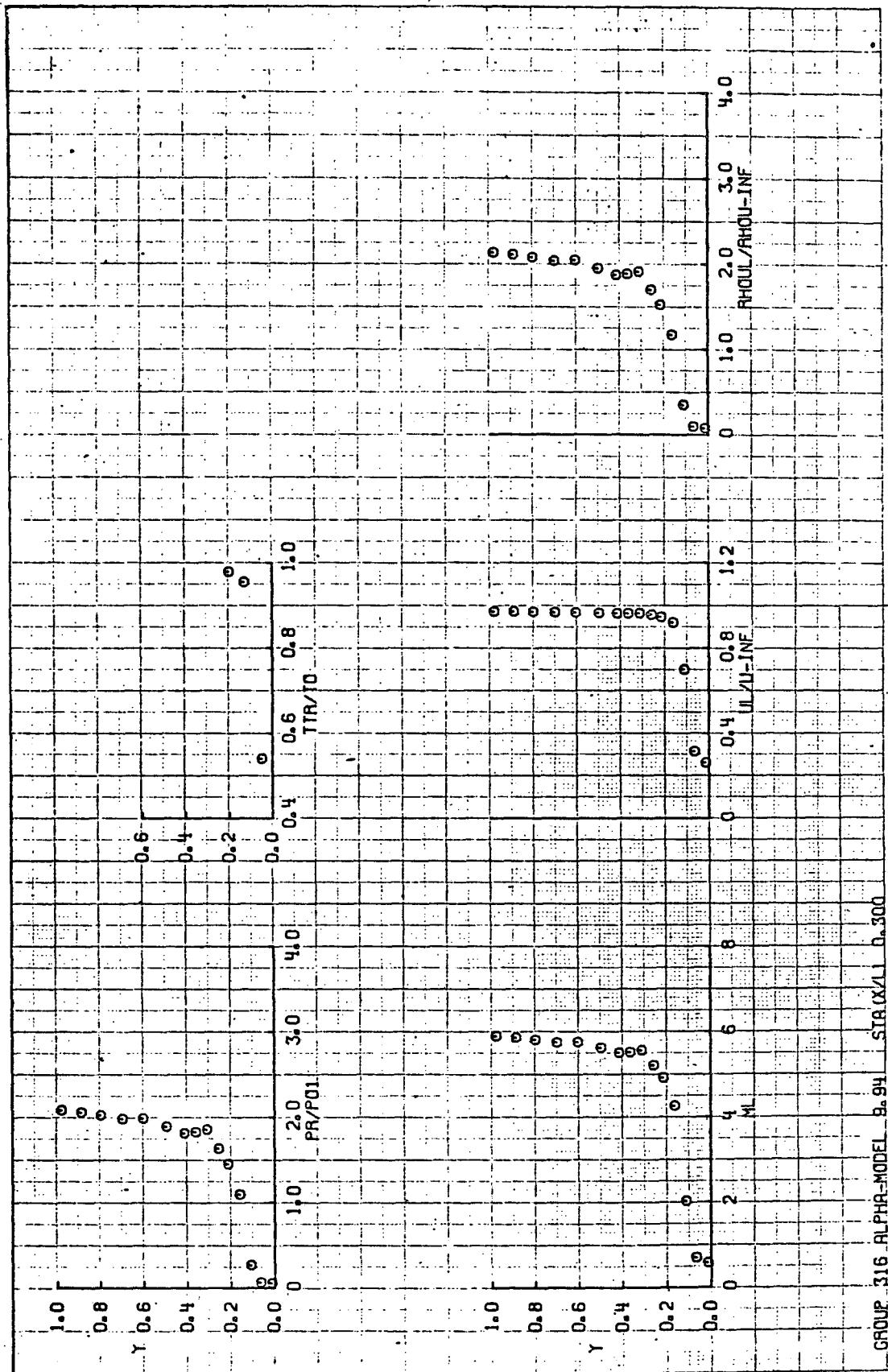
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AEIC LABS, INC. | ARNOLD AFS, TENNESSEE
YON KARMAN GAS DYNAMICS FACILITY
50 INCH HYPERSONIC TUNNEL H
VII162

Grôüp		CÔTÉG		MCDEL		MACH Nô.		P0 P5IA		T0 DÉG H		ALPHÁ-VÖDÉL		ALPHÁ-PRÉBÉND		RÖLL-MÓDEL		YAW	
316	53	NAR-DW0	8.00	862.4	1342	9.95	13.05	-23.00	-23.00	180.00	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	
T-1INF	P-1INF	P01	Q-1INF	U-1INF	RHO-1INF	MU-1INF	RE/FT	MODEL STA	L										
(DEG RI)	(P5IA)	(P5IA)	(P5IA)	(P5IA)	(P5IA)	(P5IA)	(LB-SEC/FT2)	(LEM/F13)	(FT-1)	(X/L)	(IN)								
97	8.83E-92	7.320	3.957	3867	2.652E-03	7.823E-08	3.765E-06	3.300	3.765E-06	3.300	28.90								
CW	PnS	TAP	PH	PR/P01	Y(IN)	PML/PR	ML	RFL	TL/T-1INF	RHOL/RHO-1INF	RHOUL/RHO-1INF	MUL/MU-1INF							
		(P5IA)	(P5IA)					(F1-1)											
11	1	1	1	4.2305E-01	5.7779E-02	*C14	1.960E-01	*580	3.262E-04	12.929	*261	*2948							
11	2	2	2	4.697E-01	6.418E-02	*J66	7.168E-01	*706	4.10RE-04	12.548	*313	*3038							
11	3	3	3	1.353E-01	2.668E-01	*112	1.724E-01	*2.030	2.139E-05	7.564	*698	*3518							
11	4	3	4	8.20E-01	1.096E-00	*163	4.198E-02	*4.259	1.510E-06	2.982	*919	1.1752							
11	5	3	5	1.063E-01	1.453E-00	*216	3.166E-02	*4.917	2.431E-06	2.365	*945	1.6118							
11	6	3	6	1.194E-01	1.632E-00	*258	2.819E-02	*5.214	2.980E-06	2.144	*954	1.7180							
11	7	3	7	1.388E-01	1.855E-00	*313	2.479E-02	*5.565	3.759E-06	1.918	*964	1.9149							
11	8	3	8	1.373E-01	1.826E-00	*365	1.919E-02	*5.521	3.051E-06	1.945	*9598	1.8863							
11	9	1	9	1.326E-01	1.712E-00	*415	2.538E-02	*5.499	3.600E-06	1.958	*962	1.8727							
11	10	1	10	1.377E-01	1.955E-00	*494	2.42AE-02	*5.624	3.905E-06	1.884	*965	2.0235							
11	11	1	11	1.455E-01	1.948E-00	*606	2.314E-02	*5.763	4.267E-06	1.806	*968	2.1107							
11	12	3	12	1.459E-01	1.924E-00	*701	2.322E-02	*5.751	4.235E-06	1.812	*968	2.1033							
11	13	3	13	1.492E-01	2.024E-00	*802	2.273E-02	*5.815	4.412E-06	1.777	*969	2.1445							
11	14	3	14	1.506E-01	2.054E-00	*892	2.236E-02	*5.864	4.550E-06	1.752	*970	2.1760							
11	15	3	15	1.524E-01	2.083E-00	*981	2.209E-02	*5.899	4.651E-06	1.734	*971	2.1988							

CH	TG	TTR (deg. H.)	TH/TC (deg. R.)	Y(1N)	PML/P01
1	1	724	5.395	.051	4.600E-02
2	2	120	5.938	.132	
3	3	123	6.734	.202	

GHOOLP
316



GROUP 316 ALPHA-MODEL 0.94 STAOX1 1.0 300

PAGE = 1

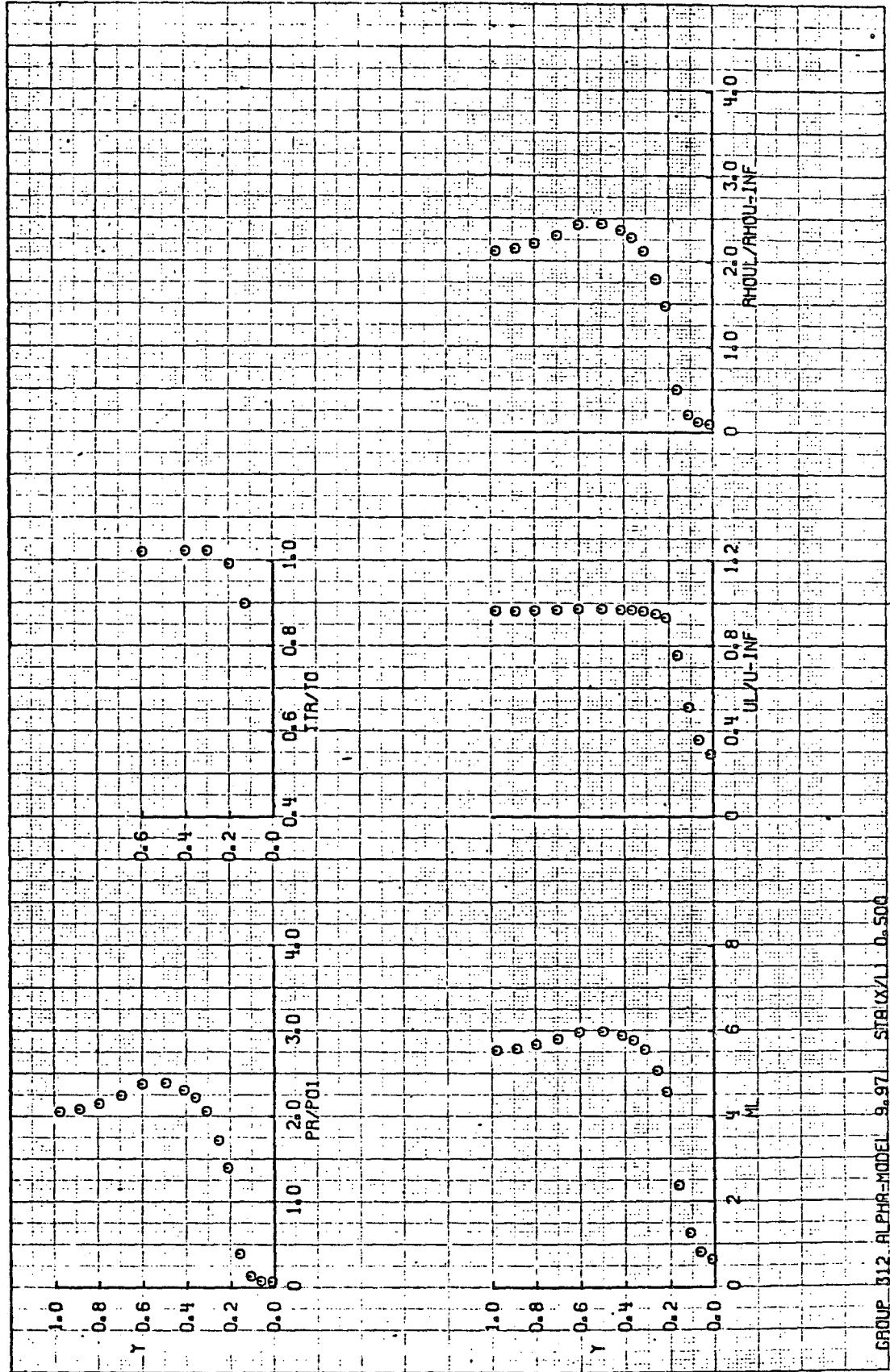
10/20/71

AEUC TAKO, INC.) ARNOLD AFS, TENNESSEE
YON KARMAN GAS DYNAMICS FACILITY
50 INCH HYPERSONIC TUNNEL H
Y11162

GROUP	CONFIG	MICHEL	MACH NO.	P0/PSIA	T0 DEG R	ALPHA-MICHEL	ALPHA-SECTOR	ALPHA-PREBEND	FOLL-MODEL	YAW
		MARDW	8.00	862.4	1341	9.97	13.03	-23.00	180.00	0
T-1NF	P-INF	P01	Q-INF	U-INF	RHO-INF	MU-INF	RE/FT	MODEL STA	L	M
(DEG R)	(PSIA)	(PSIA)	(FT/SEC)	(LB/SEC)	(LB/FT3)	(IN)	(X/L)	(IN)		
97	8.3E-02	7.320	3.957	3865	2.454E-03	7.823E-08	3.769E-06	.500	28.90	51
CR	POS	TAP	PH	PML/PR	Y(IN)	PML/PR	ML	REL	TL/T-1NF	MUL/RHO-1NF
		(PSIA)			(FT-1)					
1	3	1	5.010E-01	6.0H45E-02	.614	7.4H0E-01	.658	4.201E-04	12.701	.293
2	1	2	5.0H3E-01	6.037E-02	.666	6.370E-01	.829	5.582E-04	12.132	.361
3	3	3	9.024E-01	1.362E-01	.112	3.015E-01	1.271	1.018E-05	10.432	.513
4	3	4	2.931E-00	4.004E-01	.163	1.279E-01	2.388	3.409E-05	6.448	.758
5	3	5	1.026E-01	1.401E-00	.216	3.654E-02	4.571	2.119E-06	2.664	.933
6	3	6	1.260E-01	1.721E-00	.258	2.975E-02	5.073	3.014E-06	2.245	.950
7	3	7	1.513E-01	2.067E-00	.313	2.477E-02	5.567	4.194E-06	1.917	.964
8	3	8	1.629E-01	2.225E-00	.365	2.301E-02	5.778	4.602E-06	1.797	.968
9	3	9	1.695E-01	2.315E-00	.415	2.211E-02	5.896	5.170E-06	1.736	.971
10	3	10	1.751E-01	2.392E-00	.499	2.141E-02	5.993	5.495E-06	1.686	.973
11	3	11	1.743E-01	2.381E-00	.506	2.150E-02	5.979	5.448E-06	1.693	.973
12	3	12	1.646E-01	2.248E-00	.702	2.277E-02	5.810	4.048E-06	1.781	.969
13	3	13	1.577E-01	2.154E-00	.802	2.377E-02	5.685	4.524E-06	1.849	.966
14	3	14	1.525E-01	2.083E-00	.492	2.458E-02	5.589	4.253E-06	1.904	.964
15	3	15	1.506E-01	2.057E-00	.981	2.489E-02	5.556	4.157E-06	1.925	.963

PML/PO1

C+	TC	TH	TIR/TC	Y(IN)	PML/PO1
		(DEG R)	(DEG R)		
2	2	1206	.6993	.131	5.120E-02
3	3	1310	.9918	.202	
4	4	1372	1.0231	.303	
5	5	1372	1.0231	.402	
6	6	1369	1.0205	.599	



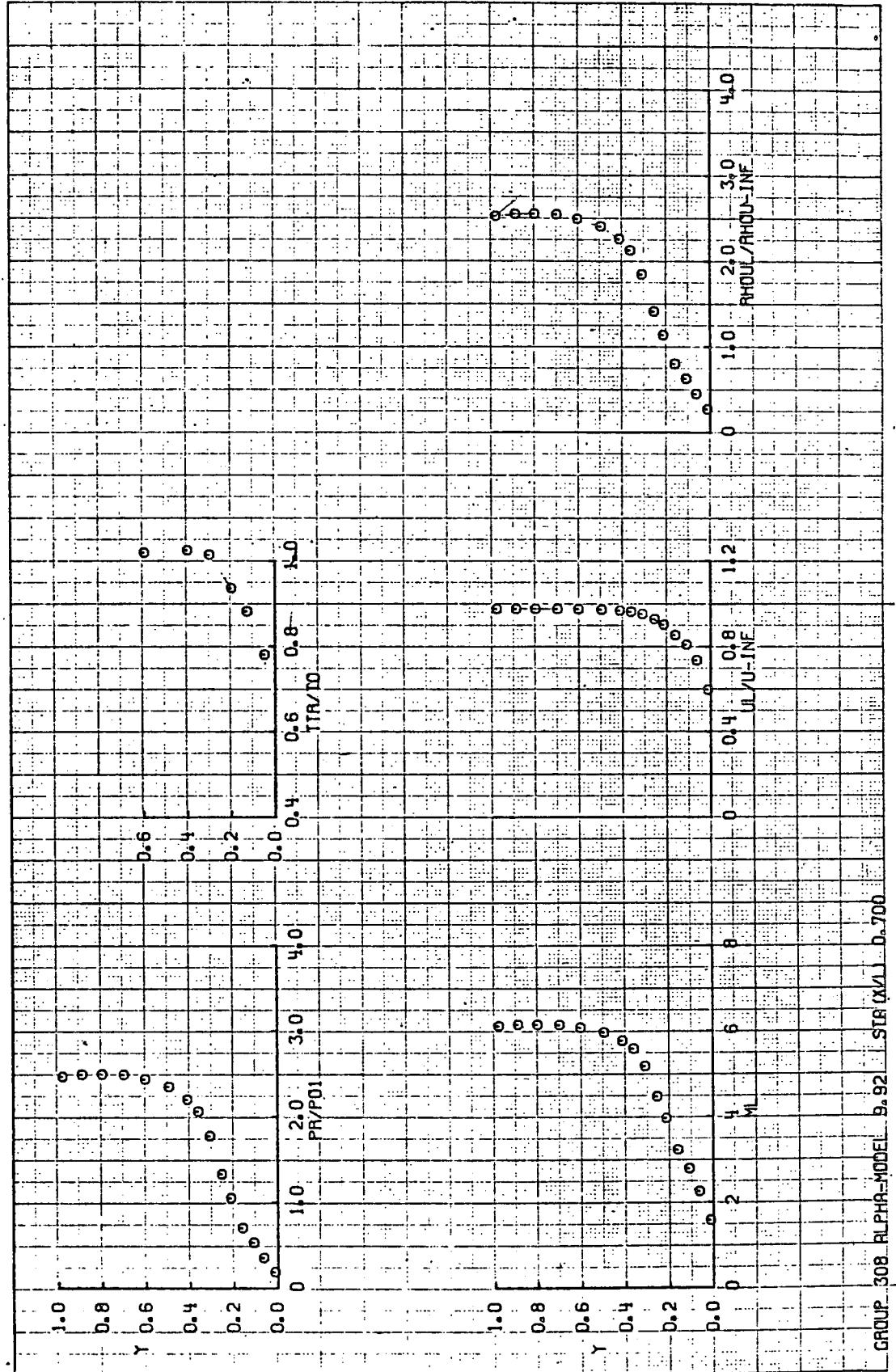
GROUP 312 ALPHA-MODEL 9.97 STRX/L - b.500

PAGE - 1

10/26/71

AEDC (RHO, INC.) ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL A
 V11162

GROUP	CONFIG	MODEL	MACH NO.	P0	P1A	T0	DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW
		NAR-DWO	8.00	860.?	1336	9.92	13.08	-23.00	181.00	0		
T-1NF	P- INF	PO1	0- INF	U- INF	RHO- INF	MU- INF	RE/FI	MODEL SIA	L			
(DEG R)	(PSIA)	(PSIA)	(PSIA)	(FT/SEC)	(LBM/FT1)	(LB-SEC/FT2)	(FT-1)	(X/L)	(IN)			
97	8.81E-02	7.301	3.947	3.858	2.456E-03	7.794E-08	3.781E-06	.700	.28490			
CH	POS	TAN	PH	PH	PH/PO1	Y(IN)	PML/PH	ML	REL	TL/T- INF	UL/U-1NF	RHOUL/RHOU- INF
			(PSIA)			(FT-1)						MUL/MU- INF
1	3	1	1.377E-00	1.986E-01	.014	2.693E-01	1.577	1.453E-05	.599			
2	3	2	2.596E-00	3.556E-01	.066	1.429E-01	2.249	2.957E-05	.736			
3	3	3	3.476E-00	5.308E-01	.112	9.570E-04	2.786	4.927E-05	.860			
4	3	4	5.139E-00	7.039E-01	.163	7.217E-02	3.224	7.274E-05	.541			
5	3	5	7.657E-00	1.049E-00	.216	4.843E-02	3.958	1.329E-06	.1339			
6	3	6	9.704E-00	1.330E-00	.258	3.820E-02	4.468	1.955E-06	.2764			
7	3	7	1.295E-01	1.774E-00	.313	2.863E-02	5.179	3.214E-06	.2173			
8	3	8	1.506E-01	2.062E-00	.365	2.463E-02	5.583	4.217E-06	.1908			
9	3	9	1.607E-01	2.201E-00	.415	2.304E-02	5.771	4.756E-06	.1802			
10	3	10	1.716E-01	2.351E-00	.499	2.161E-02	5.966	5.377E-06	.1700			
11	3	11	1.783E-01	2.435E-00	.506	2.043E-02	6.077	5.161E-06	.1645			
12	3	12	1.821E-01	2.492E-00	.702	2.034E-02	6.144	6.000E-06	.1614			
13	3	13	1.842E-01	2.494E-01	.502	2.034E-02	6.149	6.022E-06	.1612			
14	3	14	1.852E-01	2.495E-00	.892	2.036E-02	6.147	6.015E-06	.1612			
15	3	15	1.893E-01	2.462E-00	.981	2.057E-02	6.114	5.894E-06	.1625			
									.975			
CH	TC	TP	TIN/TC	Y(IN)		PML/P01						
			(DEG R)	(DEG R)								
1	1	1	1044	7814	.051							
2	2	2	1179	8825	.131							
3	3	3	1251	9364	.202							
4	4	4	1355	1.0142	.303							
5	5	5	1369	1.0247	.402							
6	6	6	1362	1.0195	.599							



PAGE = 1

10/20/71

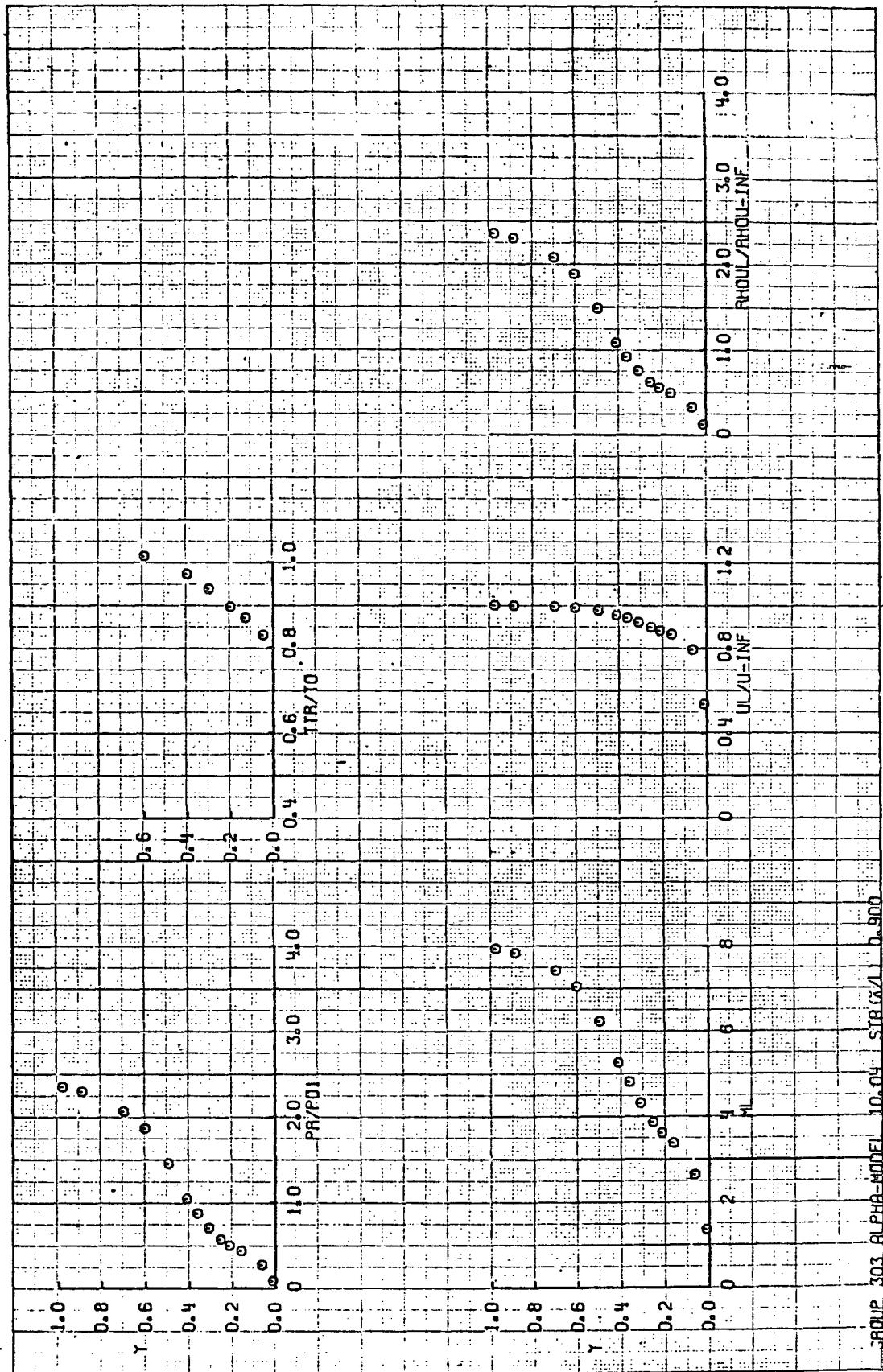
AEDC (AKO, INC.) ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL H
 Y11162

GROUP	CONFIG	MODEL	MACH NO.	P0	PSTA	T0	DEG H	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW
	303	53 HARDW0	8.00	859.9	1326	10.04	12.96	-23.00	-23.00	180.00	0.0	0
T-INF	P-INF	PO1	Q-INF	U-INF	RHO-INF	MU-INF	MU-INF	RE/FT	MODEL STA	L		
(DEG H)	(PSTA)	(PSTA)	(FT/SEC)	(LBM/FT3)	(LB-SEC/FT12)	(FT-1)	(IN)	(X/L)	(IN)			
96	8.81E+02	7.29R	3.946	384.4	2.474E+03	7.730E+08	3.822E+06	.900	.28.90			
CH	PNS	TAP	PH	PR/P01	Y(IN)	PML/PK	ML	RFL	TL/T-INF	UL/U-INF	RHOL/RHO-INF	MUL/MU-INF
1	3	1	6.120E-01	8.386E-02	4.014	3.446E-01	1.356	6.446E-04	10.0E8	539	.2374	.1279
2	3	2	2.02E-00	2.743E-01	•66	1.054E-01	2.646	2.48AE-05	5.751	793	.4164	.3303
3	4	3	3.18E-00	4.367E-01	•163	6.617E-02	3.372	4.744E-05	4.215	865	.5682	.4917
4	3	5	3.664E-00	5.021E-01	•216	5.756E-02	3.624	5.847E-05	3.805	884	.6294	.5562
5	3	6	4.166E-00	5.704E-01	•258	5.063E-02	3.870	7.126E-05	3.454	899	.6934	.6234
6	3	7	5.160E-00	7.071E-01	•313	4.087E-02	4.317	1.005E-06	2.919	922	.805	.7566
7	3	8	6.387E-00	8.751E-01	•365	3.302E-02	4.812	1.449E-06	2.451	942	.970	.9201
8	3	9	7.634E-00	1.047E-00	•415	2.761E-02	5.269	1.972E-06	2.106	956	1.1369	1.0868
9	3	10	1.065E-01	1.459E-00	•499	1.9A1E-02	6.233	3.647E-06	1.573	977	1.5221	1.4878
10	3	11	1.361E-01	1.864E-01	•606	1.550E-02	7.054	5.879E-06	1.260	990	1.9004	1.8812
11	3	12	1.509E-01	2.064E-00	•702	1.397E-02	7.431	7.233E-06	1.146	994	2.0899	2.0782
12	3	13	1.677E-01	2.294E-00	•92	1.244E-02	7.837	8.972E-06	1.039	994	2.3019	2.3019
13	3	14	1.719E-01	2.355E-00	•981	1.227E-02	7.933	9.428E-06	1.016	999	2.3575	2.3563
14	3	15	1.719E-01	2.355E-00	•981	1.227E-02	7.933	9.428E-06	1.016	999	2.3575	2.3563

CH T_C T_H T_{R/TC} Y(IN) PML/PK

	CH	T _C	T _H	T _{R/TC}	Y(IN)	PML/PK
	1	1	1102	•4310	•051	2.890E-02
	2	2	1155	•2710	•131	
	3	3	1188	•8959	•202	
	4	4	1245	•9389	•303	
	5	5	1290	•9725	•402	
	6	6	1346	1.0151	•599	

GROUP
303

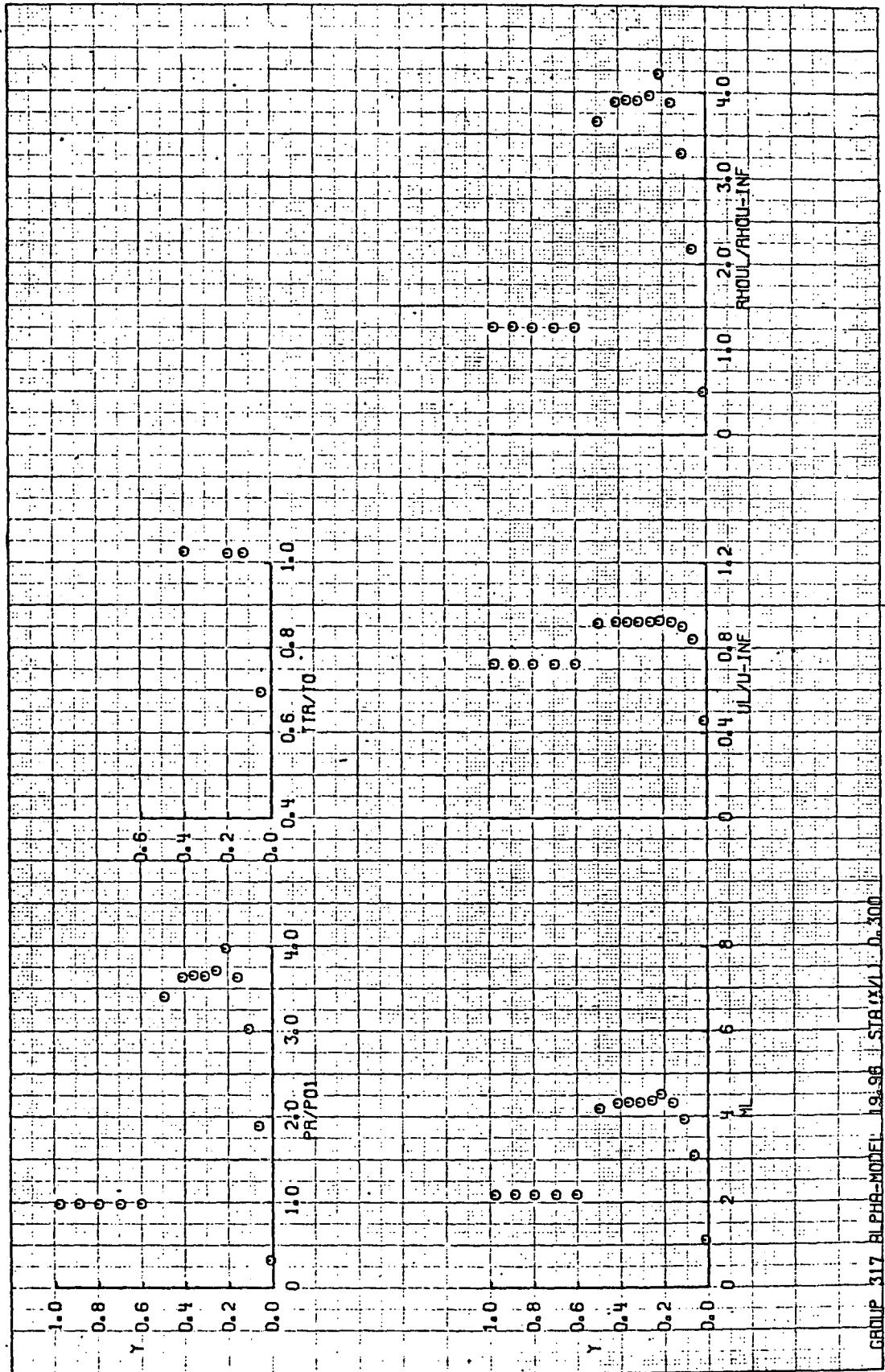


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10/20/71

NÉDÜ LÁRÓ, INC., ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL A
 V11162

GROUP	CONFIG	MODEL	MACH NO.	P0/PSTA	T0 DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW
317	53	NAR-DW0	8.00	861.4	1342	19.96	3.04	-23.00	180.00	.0
	T-1 INF	P-1 INF	P01	Q-INF	U-INF	RHO-1 INF	MU-1 INF	RE/FT	MODEL STA	L
	(DEG R)	(PSIA)	(PSIA)	(PSIA)	(FT/SEC)	(LBM/FT3)	(LB-SEC/FT2)	(FT-1)	(X/L)	(IN)
1	97	8.82E-02	7.311	3.953	3867	2.649E-03	7.829E-08	3.761E-06	.300	28.90
CH	POS	TAP	PH	PR/P01	Y(IN)	PML/PR	ML	REL	TLT-1 INF	UL/U-1 INF
			(PSIA)			(FT-1)			RHOL/RHO-1 INF	MUL/MU-1 INF
1	1	3	1	2.332E-00	3.190E-01	*614	4.639E-01	1.108	2.390E-05	1.1078
2	2	2	1.365E-01	1.895E-00	*666	7.110E-02	3.095	1.887E-06	4.733	*842
3	3	2	2.214E-01	3.028E-00	*112	4.887E-02	3.940	3.800E-06	3.361	*903
4	4	2	2.658E-01	3.623E-00	*163	4.071E-02	4.325	5.100E-06	2.910	*922
5	5	3	2.940E-01	3.976E-00	*216	3.722E-02	4.528	5.924E-06	2.705	*931
6	6	3	2.714E-01	3.712E-00	*258	3.987E-02	4.372	5.281E-06	2.861	*925
7	7	3	2.670E-01	3.652E-00	*213	4.052E-02	4.337	5.144E-06	2.898	*923
8	8	3	2.616E-01	3.658E-00	*365	4.046E-02	4.339	5.152E-06	2.896	*923
9	9	3	2.656E-01	3.633E-00	*415	4.074E-02	4.325	5.100E-06	2.910	*922
10	10	3	2.448E-01	3.404E-00	*499	4.349E-02	4.183	4.580E-06	3.067	*916
11	11	3	7.157E-00	9.790E-01	*506	1.512E-01	2.183	8.028E-05	2.067	*725
12	12	3	7.139E-00	9.765E-01	*702	1.516E-01	2.179	7.997E-05	7.079	*725
13	13	3	7.143E-00	9.771E-01	*502	1.515E-01	2.179	7.997E-05	7.079	*725
14	14	3	7.161E-00	9.622E-01	*92	1.507E-01	2.187	8.060E-05	7.055	*726
15	15	3	7.152E-00	9.782E-01	*481	1.513E-01	2.181	8.012E-05	7.073	*725
									1.7339	1.2571
CH	TC	TR	TR/TC	Y(IN)	PML/PR					
		(DEG R)	(DEG R)							
1	1	914	.6960		.051					
2	2	1373	.10231		.131					
3	3	1370	.10205		.202					
4	4	1376	.10253		.402					



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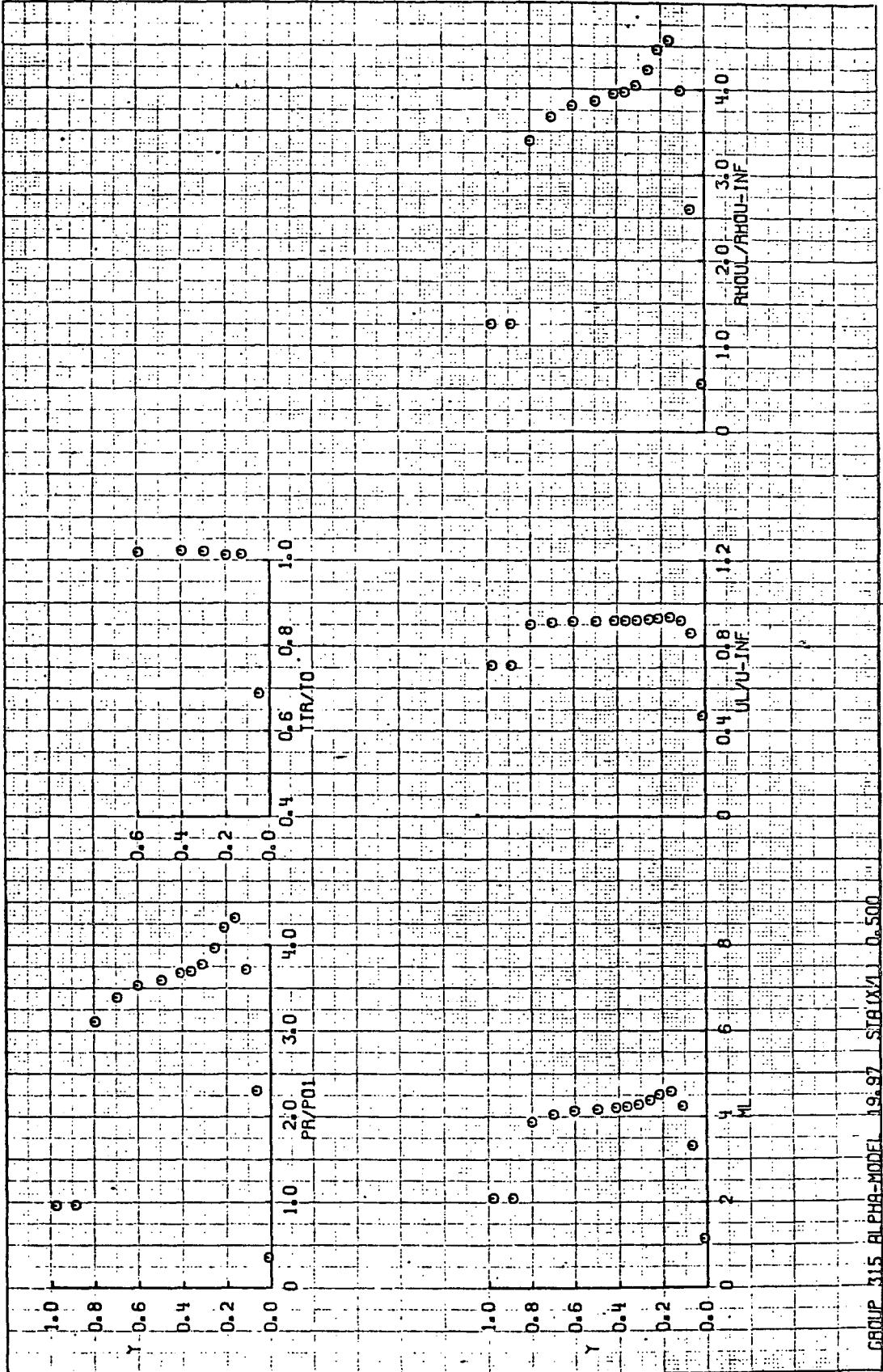
10/20/71

AEDC (AHO, INC.) ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL B
 Y11162

GROUP	CONFIG	MACH NO.	P0	PSIA	T0 DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW
	215	2.3	NAR-DW0	8.00	860.5	1.341	19.97	3.03	-23.00	180.00
	I=INF	P-INF	P01	0-INF	U-INF	RHO-INF	MU-INF	REL	MODEL SIA	L
	(DEG R)	(PSIA)	(PSIA)	(FT/SEC)	(LBM/FT3)	(LB-SEC/FT2)	(FT-1)	(X/L)	(IN)	
	9.7	8.81E-02	7.303	3.948	.38E5	2.49E-03	7.023E-08	3.761E-06	.500	28.90
CH	PNS	TAP	PH	PR/P01	Y1(IN)	RHO/PM	ML	REL	TL/T-INF	UL/U-INF
	(PSIA)		(PSIA)			(FT-1)			RHO/L/RHO-L-INF	RHO/L/RHO-U-INF
1	1	1	2.589E-00	3.545E-01	.014	4.429E-01	1.147	2.667E-05	1.1910	7.963
2	2	2	1.688E-01	2.307E-00	.066	6.806E-02	3.323	2.438E-06	.862	2.6063
3	3	3	2.717E-01	3.720E-00	.112	4.221E-02	4.247	5.102E-06	.919	3.9913
4	4	3.157E-01	4.323E-00	.163	3.632E-02	4.585	6.548E-06	2.652	.933	4.3636
5	5	3.074E-01	4.209E-00	.216	3.731E-02	4.522	6.257E-06	2.711	.931	4.9063
6	6	6	2.844E-01	3.964E-00	.258	3.955E-02	4.390	5.675E-06	.843	4.7920
7	7	7	2.760E-01	3.779E-00	.313	4.155E-02	4.282	5.239E-06	.921	4.5758
8	8	8	2.701E-01	3.698E-00	.365	4.246E-02	4.233	5.049E-06	.918	4.4002
9	9	9	2.685E-01	3.676E-00	.415	4.271E-02	4.222	5.005E-06	.918	4.3218
10	10	10	2.624E-01	3.502E-00	.499	4.371E-02	4.173	4.923E-06	.915	3.9488
11	11	11	2.571E-01	3.528E-00	.506	4.450E-02	4.134	4.682E-06	.913	4.2258
12	12	12	2.477E-01	3.395E-00	.702	4.625E-02	4.025	4.124	.913	3.8680
13	13	13	2.270E-01	3.104E-00	.02	5.052E-02	3.874	3.926E-06	.909	3.8039
14	14	14	7.038E-00	9.637E-01	.692	1.629E-01	2.095	7.190E-05	.7350	.6744
15	15	15	6.991E-00	9.581E-01	.981	1.639E-01	2.089	7.43E-05	.709	3.139
										3.3929
										3.335
										6.067
										1.2566
										1.2615
										6.079

CH TC TH(TC) Y1(IN) PML/P01

	(DEG, H)	(DEG, R)	Y1(IN)	PML/P01
1	1	92.1	.6893	.051
2	2	1.1360	.0142	1.570E-01
3	3	1.1357	.0119	.131
4	4	1.1369	.0209	.202
5	5	1.1370	.0216	.303
6	6	1.1363	.0119	.599

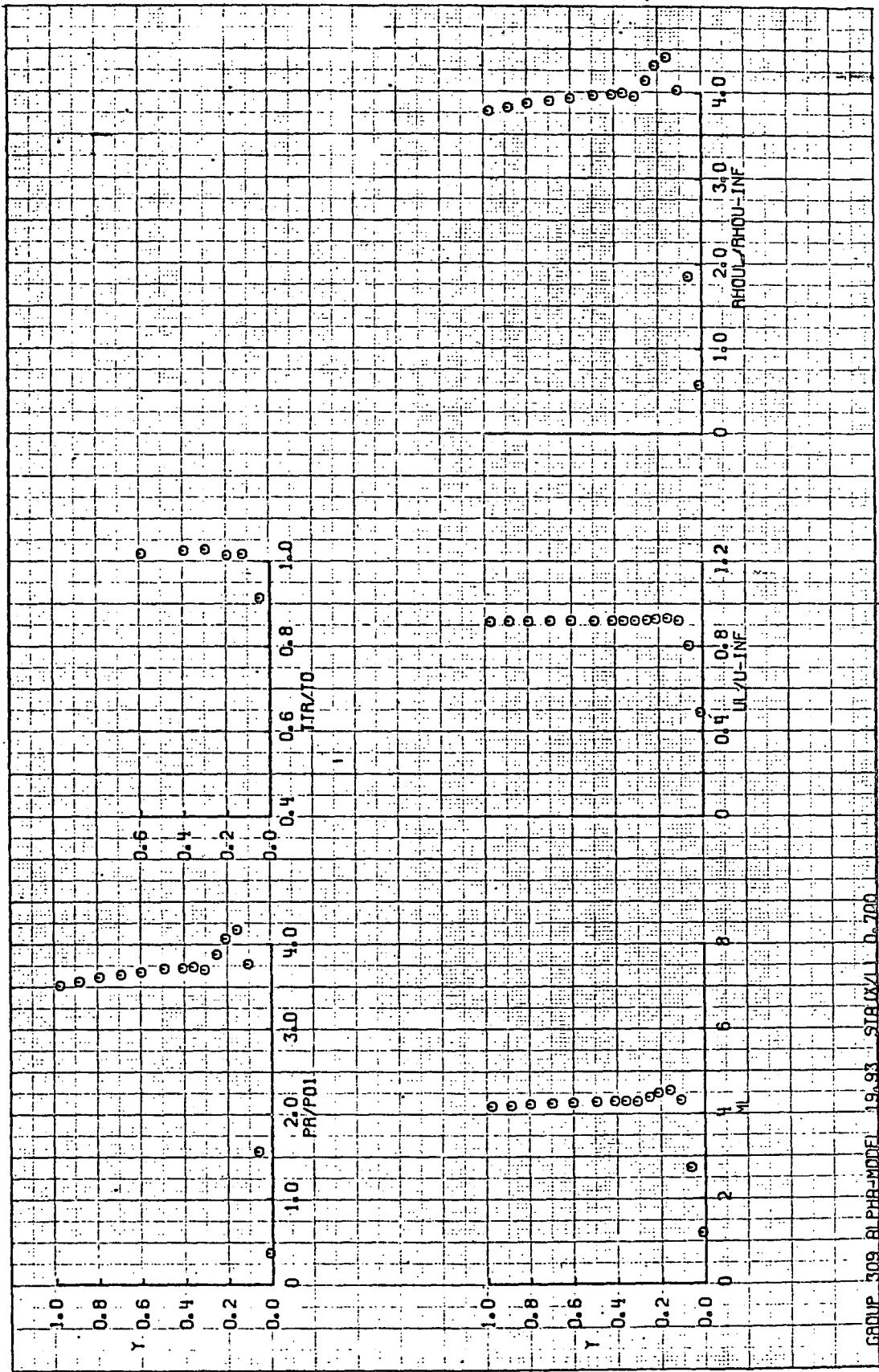


PAGE = 1

10/26/71

KEDIC TARDI, INC.) ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL H
 V11162

GROUP	CONFIG	MODEL	MACH NO.	P0 PSIA	T0 DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW
309	53	AARDW	8.00	860.6	1.337	19.93	3.07	-23.00	180.00	0.0
	T-INF	P-INF	P01	0-INF		RHO-INF	MU-INF	RE/FI	MODEL_SIA	L
	(DEG R)	(PSIA)	(PSIA)	(FT/SEC)		(LBM/FT3)	(LB-SEC/FT2)	(FT-1)	(IN/L)	
	97	8.81E-12	7.304	3.949		2.45E-13	7.800E-08	3.778E-06	1.00	28.90
	CH	POS	TAP	PR	PR/P01	Y(IN)	PML/PR	ML	HEL	TL/I-1NF
			(PSIA)			(FT-1)			RHO/L/RHO=INF	MUL/MU-INF
	1	3	1	2.682E-00	3.672E-01	4.194E-01	1.192	2.781E-05	1.0765	4.89
	2	3	2	1.136E-01	1.556E-00	9.899E-02	2.733	1.425E-06	5.533	.804
	3	3	3	2.752E-01	3.764E-00	4.087E-02	4.317	5.298E-06	2.919	.922
	4	3	4	3.045E-01	4.163E-00	4.694E-02	4.546	6.227E-06	6.688	.932
	5	3	5	2.968E-01	4.0163E-00	2.16	3.790E-02	4.487	6.008E-06	2.745
	6	3	6	2.834E-01	3.349E-00	2.58	3.970E-02	4.392	5.559E-06	2.851
	7	3	7	2.790E-01	3.696E-00	313	4.157E-02	4.274	5.131E-06	.925
	8	3	8	2.726E-01	3.732E-00	365	4.127E-02	4.295	5.214E-06	.920
	9	3	9	2.715E-01	3.717E-00	415	4.143E-02	4.288	5.183E-06	.921
	10	3	10	2.707E-01	3.706E-00	449	4.156E-02	4.280	5.153E-06	.920
	11	3	11	2.679E-01	3.668E-00	506	4.198E-02	4.259	5.071E-06	.919
	12	3	12	2.657E-01	3.638E-00	702	4.234E-02	4.241	5.004E-06	.919
	13	3	13	2.637E-01	3.611E-00	902	4.265E-02	4.224	4.938E-06	.918
	14	3	14	2.596E-01	3.554E-00	992	4.333E-02	4.190	4.815E-06	.909
	15	3	15	2.506E-01	3.513E-00	981	4.384E-02	4.165	4.723E-06	.908
	CH	TC	TIR	TIR/TC	Y(IN)	PML/P01				
			(DEG R)	(DEG R)						
"	1	1	1222	.9140	.051	1.540E-01				
"	2	2	1360	1.0172	.131					
"	3	3	1356	1.0142	.202					
"	4	4	1374	1.0277	.303					
"	5	5	1371	1.0254	.402					
"	6	6	1361	1.0180	.598					
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42

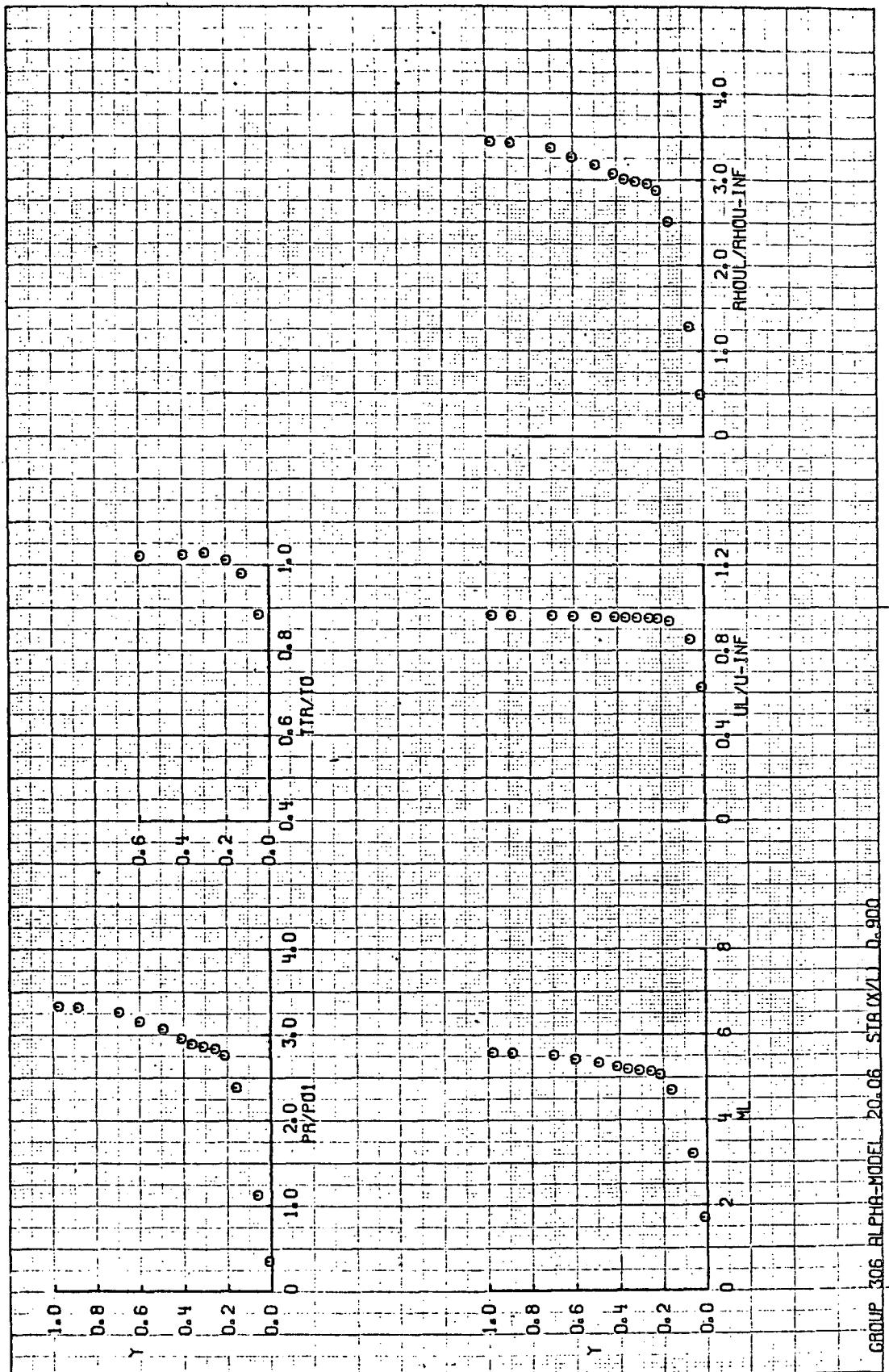
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10/20/71

AEDC TARO, INC., ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL A
 V1162

GRD(JP)	CONFIG	MODEL	MACH NO.	P0	PSTA	10 DEG R	ALPHA-MODEL	ALPHA-SECTOR	ROLL-MODEL	YAW
306	53	NAR-DWO	8.00	86.3	1334	20.06	2.94	-23.00	180.00	0.0
T- INF	P- INF	P01	G-INF	U-INF	RHO-INF	MU-INF	RE/FT	MODEL STA	L	
(DEG Q)	(PSIA)	(PSIA)	(PSIA)	(FT/SEC)	(LB/SEC/F13)	(LB-SEC/F12)	(FT/L)	(X/L)	(IN)	
97	8.81E-02	7.302	3.948	3.855	2.460E-03	7.783E-08	3.790E-06	.900	28.90	
CH	POS	TAP	PH	PR/P01	Y(IN)	PML/PH	ML	RFL	TL/T-INF	UL/U-INF
			(PSIA)			(FT-1)	(FT-1)			
1	1	2.530E 00	3.466E-01	.014	2.378E-01	1.696	2.697E .05	.8759	.628	.7795
2	2	8.197E 00	1.123E 00	.666	7.340E-02	3.196	1.155E .06	4.535	.851	1.5057
3	3	1.141E 01	2.384E 00	.163	3.456E-02	4.702	3.766E .06	2.545	.938	2.6827
4	4	2.014E 01	2.759E 00	.216	2.987E-02	5.063	4.850E .06	2.252	.950	3.0319
5	5	2.014E 01	2.434E 00	.258	2.904E-02	5.139	5.192E .06	2.198	.952	3.1068
6	6	2.012E 01	2.434E 00	.313	2.679E-02	5.159	5.177E .06	2.182	.953	3.1287
7	7	2.093E 01	2.662E 00	.365	2.849E-02	5.187	5.274E .06	2.163	.954	3.1567
8	8	2.112E 01	2.893E 00	.415	2.786E-02	5.245	5.487E .06	2.122	.955	3.2172
9	9	2.159E 01	2.957E 00	.499	2.697E-02	5.343	5.857E .06	2.057	.958	3.3195
10	10	2.234E 01	3.067E 00	.606	2.613E-02	5.419	6.160E .06	2.008	.960	3.4006
11	11	2.303E 01	3.154E 00	.702	2.265E-02	5.515	6.359E .06	1.949	.962	3.5042
12	12	2.349E 01	3.265E 00	.792	2.224E-02	5.515	6.359E .06	1.949	.963	3.5556
13	13	2.425E 01	3.321E 00	.992	2.441E-02	5.562	6.762E .06	1.920	.964	3.5663
14	14	2.425E 01	3.331E 00	.981	2.474E-02	5.571	6.805E .06	1.915	.964	3.4369
15	15	2.425E 01	3.331E 00							

CH	TC	TH	TH	Y(IN)	PML/PH
	(DEG R)	(DEG R)	(DEG R)		
1	1	1.180	.8846	.051	8.240E-02
2	2	1.307	.9798	.131	
3	3	1.349	.0112	.202	
4	4	1.372	.0285	.303	
5	5	1.366	.0240	.402	
6	6	1.362	.0210	.599	



PAGE = 1

10/20/71

AEDC (AKO, INC.) ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL R
 VIII162

GROUP	CONFIG	MODEL	MACH NO.	P0	PSIA	10 DEG R	ALPHA-MONEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAN
31A	53	MARDMO	8.00	859.3	1341	30.00	-7.00	-23.00	-180.00	-20	
CH	POS	TAP	PH	PR	PR/PO1	Y(IN)	PML/PR	ML	REF	TL/T-INF	UL/U-INF
		(PSIA)				(FT-1)				RHO/RHO-1NF	RHOUL/RHOU-1NF
1	1	1	6.316E-00	84.661E-01	.4014	3.579E-01	1.325	6.567E-05	.529	2.5152	1.3317
2	2	2	2.287E-01	3.0136E-00	.066	9.885E-02	2.735	2.858E-06	.5528	.804	4.6469
3	3	3	3.063E-01	4.1200E-00	.4112	7.380E-02	3.187	4.276E-06	.4553	.850	5.6416
4	4	4	3.045E-01	4.175E-00	.163	7.425E-02	3.177	4.240E-06	.4572	.849	5.685
5	5	5	2.456E-01	4.053E-00	.216	7.648E-02	3.128	4.064E-06	.4667	.845	5.5039
6	6	6	2.718E-01	3.6727E-00	.258	8.317E-02	2.995	3.615E-06	.4939	.832	5.2011
7	7	7	2.617E-01	3.588E-00	.313	H.639E-02	2.937	3.431E-06	.5065	.826	5.0717
8	8	8	2.614E-01	3.584E-00	.365	8.649E-02	2.935	3.425E-06	.5069	.826	4.1857
9	9	9	2.599E-01	3.4563E-00	.415	8.699E-02	2.925	3.395E-06	.5091	.825	5.0675
10	10	10	2.555E-01	3.4503E-00	.499	8.849E-02	2.899	3.319E-06	.5147	.822	4.9911
11	11	11	6.670E-00	9.145E-01	.506	3.390E-01	1.372	6.948E-05	.1025	.543	2.5623
12	12	12	6.660E-00	9.0132E-01	.702	3.395E-01	1.370	6.931E-05	.1033	.543	2.5603
13	13	13	6.564E-00	9.0000E-01	.702	3.444E-01	1.358	6.835E-05	.10080	.539	2.5484
14	14	14	6.521E-00	8.941E-01	.992	3.467E-01	1.353	6.787E-05	.10103	.537	2.5425
15	15	15	6.514E-00	8.937E-01	.981	3.469E-01	1.351	6.771E-05	.10111	.537	2.5405

PML/PO1

ML

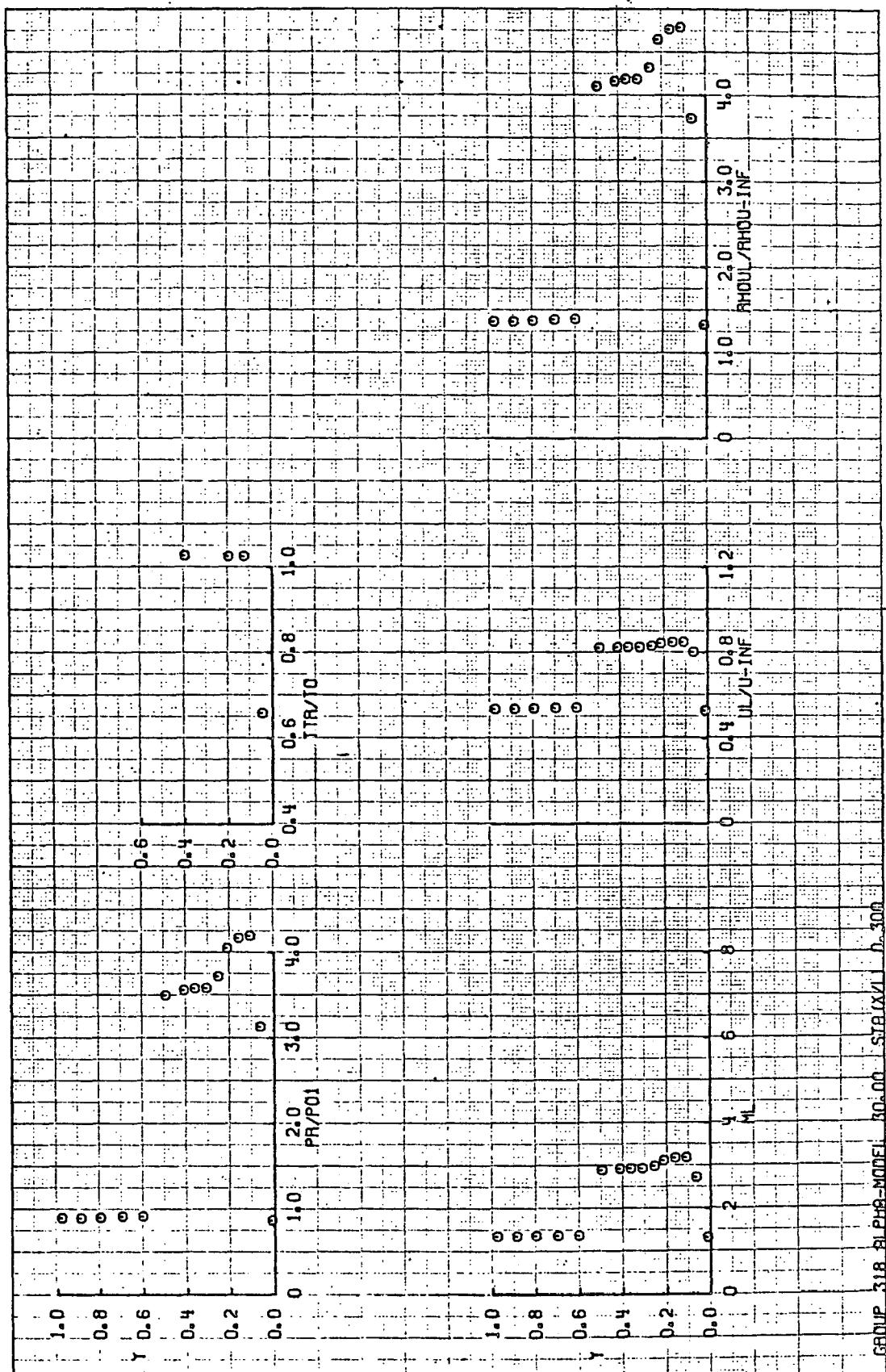
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RHO/RHO-1NF

RHOUL/RHOU-1NF



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10/20/71

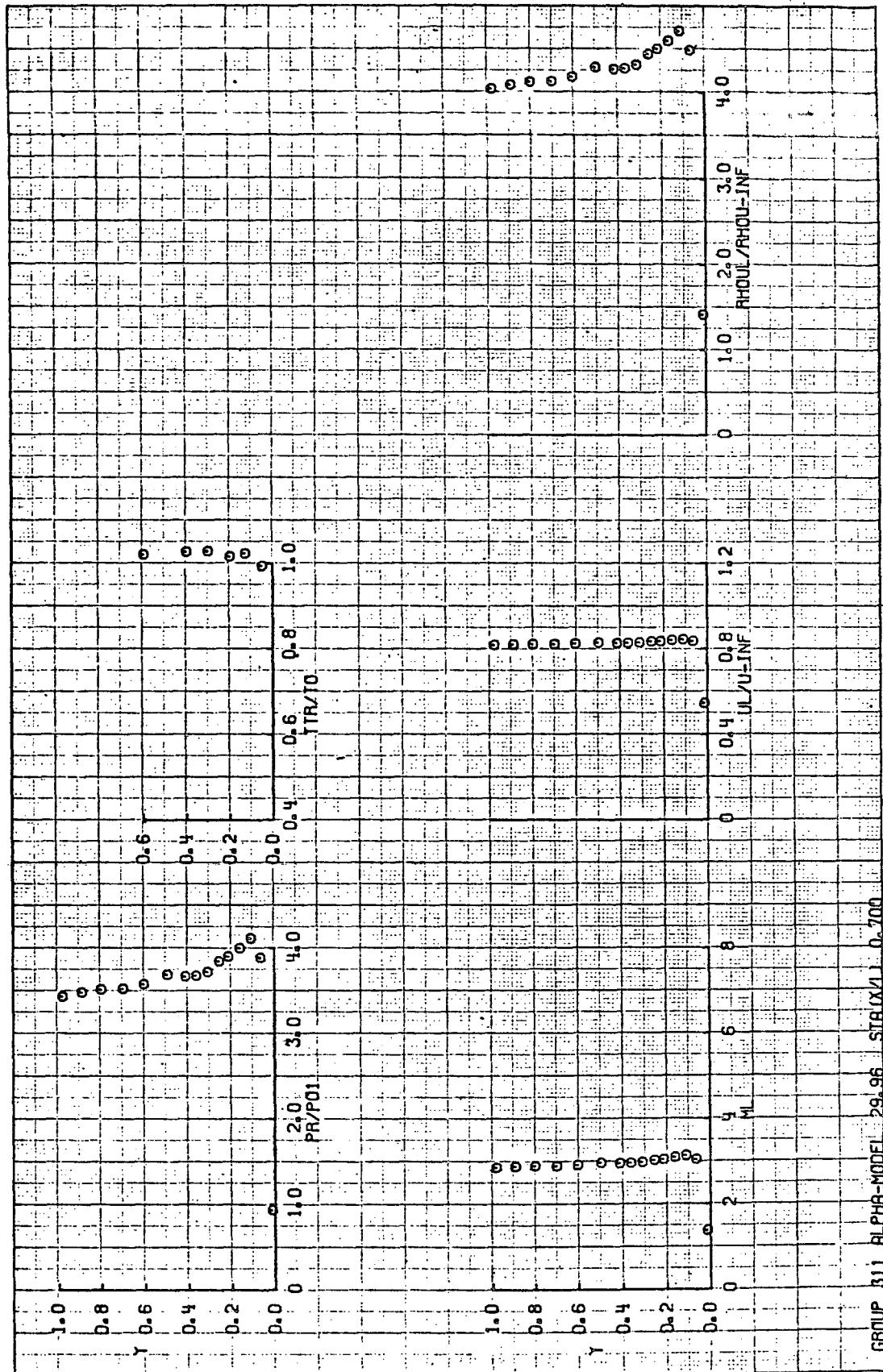
AÉC TAKO, INC.) ARNOLD AFS, TENNESSEE
VON KARWAN GAS DYNAMICS FACILITY

50 INCH HYPERSONIC TUNNEL A

Y11162

GROUP	CONFIG	MODEL	WACH NO.	P0 PSIA	T0 DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBENO	ROLL-MODEL	YAW
311	53	NAR-DW0	8.00	862.3	1341	29.97	-6.97	-23.00	180.00	0
	T-1NF	P-INF	PSIA	0-1NF	U-1NF	RHO-1NF	MU-1NF	RE/FT	MODEL STA	L
	(DF6 R)	(PSIA)	(PSIA)	(FT/SEC)	(FT/SEC)	(LEM/FT3)	(LB-SEC/FT2)	(FT-1)	(X/L)	(IN)
	97	8.83E-02	7.319	3.957	3865	2.453E-03	7.823E-08	3.769E-06	.700	28.90
CH	PNS	TAP	PR	PR/PO1	Y(IN)	PML/PB	ML	REFL	IL/T-1NF	UL/U-1NF
						(FT-1)			RHOL/RHO-1NF	RHOU/RHO-1NF
				(PSIA)						
1	3	1	6.829E-00	9.332E-01	*.014	3.343E-01	1.384	7.116E-05	9.978	*.546
2	3	2	2.840E-01	3.848E-00	*.066	8.024E-02	3.052	3.839E-06	4.821	*.838
3	3	3	3.912E-01	4.116E-00	*.112	7.580E-02	3.144	4.160E-06	4.636	*.846
4	3	4	2.924E-01	4.000E-00	*.163	7.799E-02	3.097	3.994E-06	4.729	*.842
5	3	5	2.956E-01	3.402E-00	*.216	7.996E-02	3.058	3.859E-06	4.809	*.838
6	3	6	2.811E-01	3.841E-00	*.258	8.122E-02	3.032	3.773E-06	4.861	*.836
7	3	7	2.725E-01	3.723E-00	*.313	8.393E-02	2.983	3.614E-06	4.964	*.831
8	3	8	2.690E-01	3.676E-00	*.365	8.488E-02	2.964	3.551E-06	5.006	*.829
9	3	9	2.685E-01	3.669E-00	*.415	8.503E-02	2.960	3.539E-06	5.014	*.829
10	3	10	2.700E-01	3.689E-00	*.499	8.458E-02	2.970	3.570E-06	4.993	*.830
11	3	11	2.641E-01	3.577E-00	*.506	8.722E-02	2.921	3.417E-06	5.099	*.825
12	3	12	2.561E-01	3.527E-00	*.702	8.846E-02	2.899	3.352E-06	5.147	*.822
13	3	13	2.574E-01	3.517E-00	*.02	8.871E-02	2.896	3.340E-06	5.155	*.822
14	3	14	2.550E-01	3.484E-00	*.92	8.955E-02	2.882	3.299E-06	5.186	*.820
15	3	15	2.514E-01	3.435E-00	*.981	9.002E-02	2.860	3.236E-06	5.235	*.818
										4.9320
										4.0407
										4.746
CH	TC	TR	TR/TC	Y(IN)		FML/P01				
		(DEG R)	(DEG R)							
n	1	1	1.332	*.9932	*.051	3.12E-01				
n	2	2	1.372	1.0231	*.131					
n	3	3	1.362	1.0157	*.202					
n	4	4	1.379	1.0291	*.302					
n	5	5	1.377	1.0268	*.402					
n	6	6	1.368	1.0291	*.599					

F7

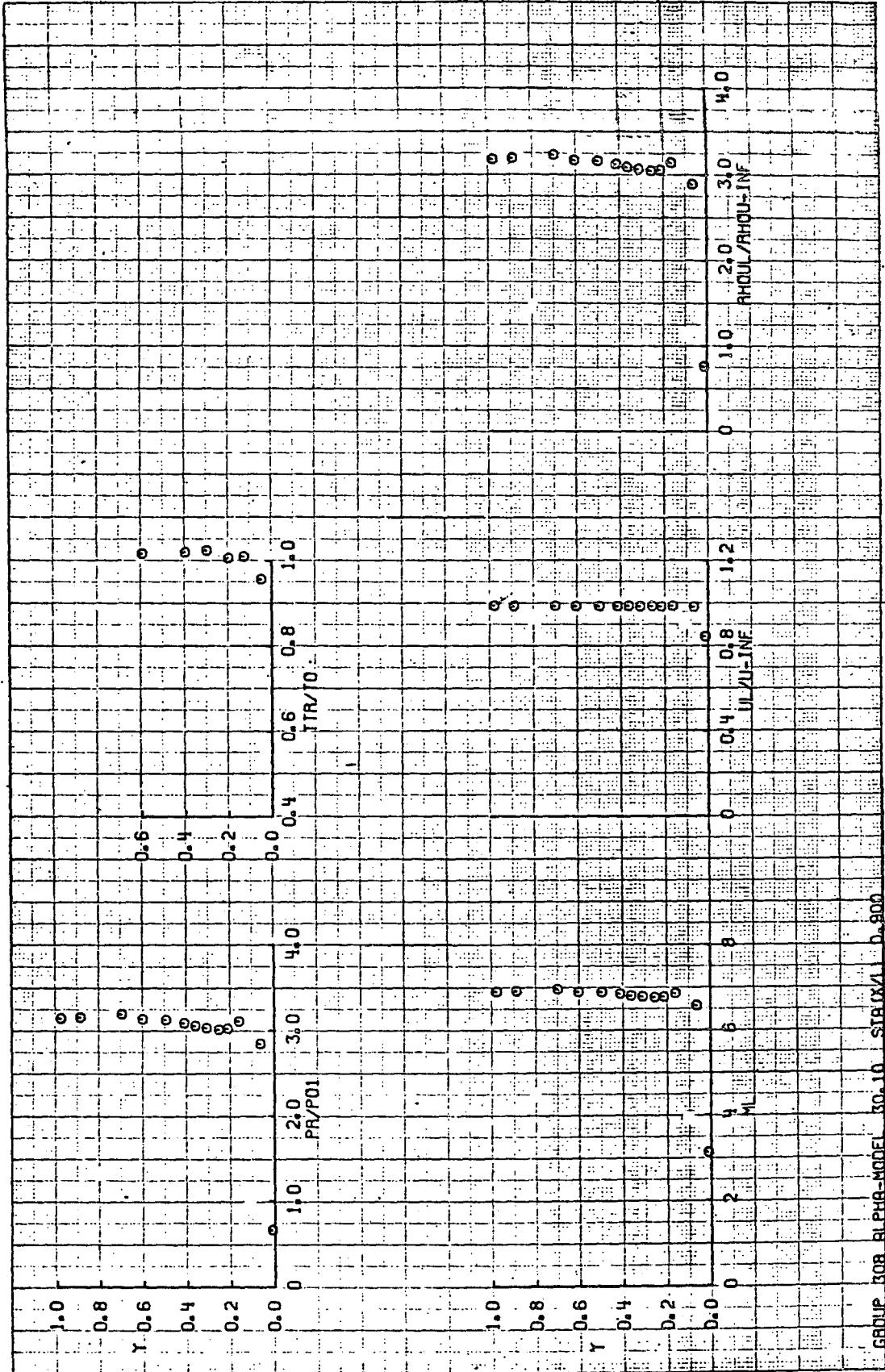


AEDC (AKO), INC.) ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL R
 V11162

GROUP	CONFIG	MODEL	MACH NO.	PO/PSIA	TO DEG R	ALPHA-MODEL	ALPHA-SECTOR	ALPHA-PREBEND	ROLL-MODEL	YAW	-23.00	180.00	.0	
30A	53	KAR-DIO	8.00	860.7	1336	30.10	-7.10							
1	1	4.451E-01	6.641E-01	.0014	7.650E-02	3.0128	6.702E-05	4.6667	.845	.9019	7.7619	4.301		
	2	2.082E-01	2.499E-00	.0066	1.783E-02	6.573	7.673E-06	1.431	.983	2.9410	2.8914	1.409		
	3	7.537E-01	1.032E-00	.0112	4.924E-02	3.925	1.295E-06	3.382	.902					
	4	2.270E-01	3.108E-00	.0163	1.035E-02	6.806	9.194E-06	1.323	.987	3.1812	3.1411	1.293		
	5	2.219E-01	3.025E-00	.0216	1.679E-02	6.774	8.724E-06	1.356	.986	3.1048	3.0617	1.328		
	6	2.149E-01	3.011E-00	.0258	1.671E-02	6.759	8.646E-06	1.361	.986	3.0919	3.0483	1.334		
	7	2.215E-01	3.032E-00	.0313	1.676E-02	6.782	8.763E-06	1.353	.986	3.1113	3.0684	1.325		
	8	2.232E-01	3.055E-00	.0365	1.663E-02	6.808	8.891E-06	1.344	.987	3.1323	3.0903	1.315		
	9	2.257E-01	3.090E-00	.0415	1.644E-02	6.847	9.092E-06	1.330	.987	3.1649	3.1242	1.300		
	10	2.205E-01	3.129E-00	.0499	1.624E-02	6.890	9.317E-06	1.315	.988	3.2009	3.1616	1.284		
	11	2.292E-01	3.134E-00	.0506	1.619E-02	6.899	9.369E-06	1.312	.988	3.2091	3.1701	1.280		
	12	2.336E-01	3.149E-00	.0502	1.598E-02	6.966	9.727E-06	1.299	.989	3.2653	3.2285	1.256		
	13	9.874E-01	1.352E-00	.0502	3.758E-02	4.507	2.013E-06	2.726	.930	1.5642	1.4365	2.700		
	14	2.308E-01	3.159E-00	.0492	1.604E-02	6.925	9.504E-06	1.303	.988	3.2305	3.1924	1.271		
	15	2.298E-01	3.146E-00	.0481	1.615E-02	6.909	9.421E-06	1.308	.988	3.2173	3.1787	1.276		

CW TC TTH TTH/P01 Y (IN) PML/P01

CW	TC	TTH	TTH/M	Y (IN)	(DEG R)	PML/R	RHO-1 INF (LEM/FT3)	MU-1 INF (LB-SEC/FT2)	RE/FT (FT-1)	MODEL STA (X/L)	L (IN)	RHO/U-1 INF RHO/L/RHO-1 INF	MUL/MU-1 INF
1	1	1278	.9566	.051	5.080E-02								
2	2	1349	1.0097	.131									
3	3	1344	1.0060	.202									
4	4	1367	1.0232	.303									
5	5	1362	1.0195	.402									
6	6	1357	1.0157	.599									



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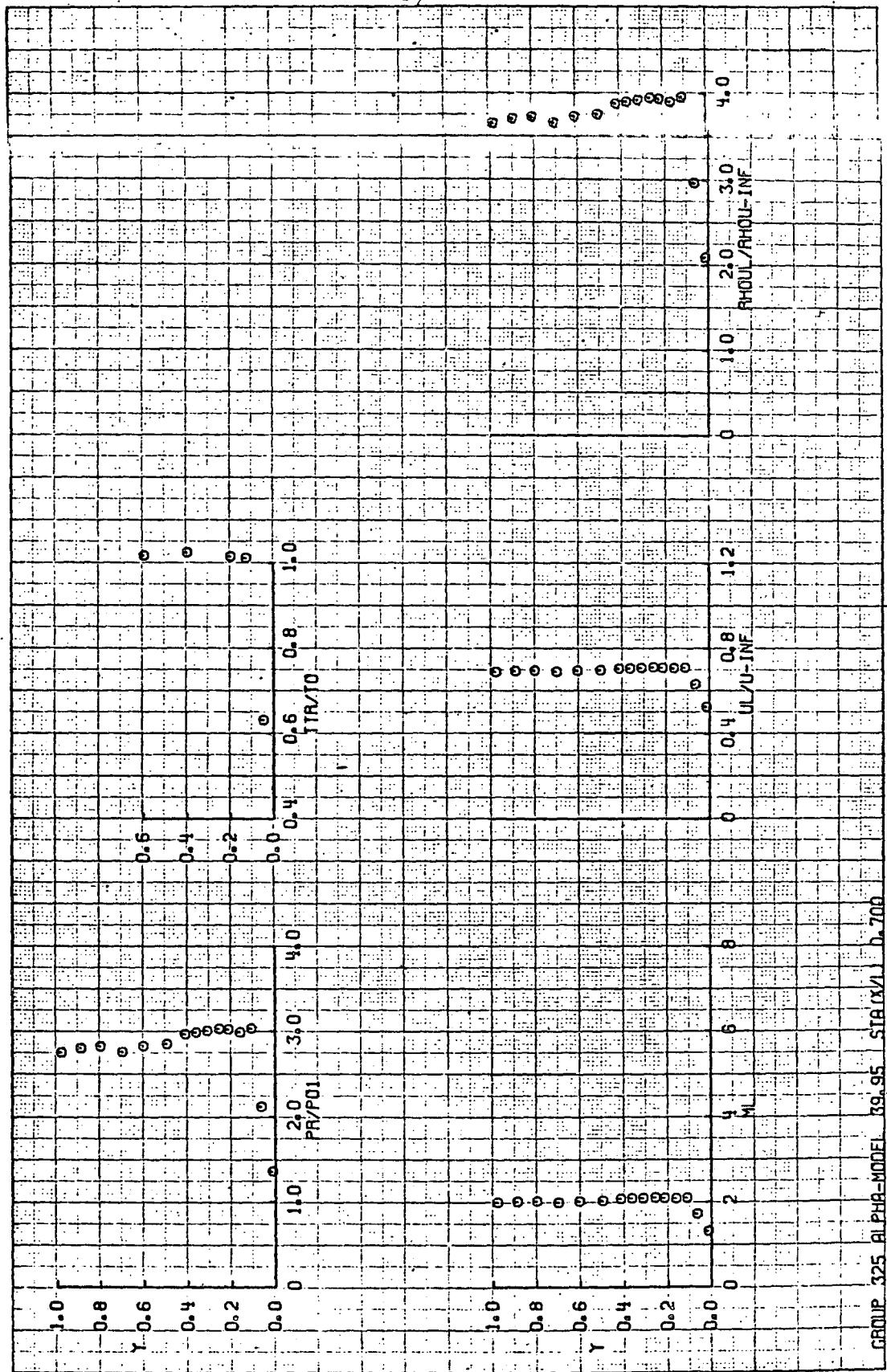
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AEDC (TNO) INC., ARNOLD AFS, TENNESSEE
 YON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL A
 V11162

GROUP	CONFIG	MODEL	MACH NO.	P0 PSIA	10 DEG R	ALPHA-MODEL	ALPHA-SECTOR	ROLL-MODEL	YAW
325	5.3	NAR-D#0	8.00	801.1	1339	39.95	10.05	-50.00	180.00
	T- INF	P- INF	P01	0- INF	U- INF	RHO- INF	MU- INF	RE/FT	MODEL STA
	(DEG R)	(PSIA)	(PSIA)	(FT/SEC)	(FT)	(LB/FT3)	(LB-SEC/FT2)	(FT-1)	L
0	1	3	1	9.93E-00	1.359E-00	.014	3.619E-01	1.315	1.034E-06
0	2	3	2	1.548E-01	2.118E-00	.066	2.323E-01	1.718	1.643E-06
0	3	3	3	2.216E-01	3.033E-00	.112	1.622E-01	2.101	2.462E-06
0	4	3	4	2.182E-01	2.945E-00	.163	1.648E-01	2.081	2.113E-06
0	5	3	5	2.209E-01	3.021E-00	.216	1.629E-01	2.095	2.647E-06
0	6	3	6	2.211E-01	3.025E-00	.258	1.626E-01	2.097	2.652E-06
0	7	3	7	2.196E-01	3.005E-00	.313	1.637E-01	2.089	2.433E-06
0	8	3	8	1.180E-01	2.983E-00	.365	1.649E-01	2.081	2.113E-06
0	9	3	9	2.166E-01	2.964E-00	.415	1.660E-01	2.073	2.394E-06
0	10	3	10	2.082E-01	2.948E-00	.499	1.727E-01	2.028	2.286E-06
0	11	3	11	2.062E-01	2.921E-00	.606	1.744E-01	2.019	2.263E-06
0	12	3	12	2.013E-01	2.755E-00	.702	1.786E-01	1.991	2.200E-06
0	13	3	13	2.060E-01	2.318E-00	.602	1.746E-01	2.017	2.258E-06
0	14	3	14	2.044E-01	2.197E-00	.992	1.759E-01	2.009	2.240E-06
0	15	3	15	2.000E-01	2.748E-00	.981	1.790E-01	1.989	2.195E-06

C#	TC	TTR	TTR/TC	Y (IN)	PML/P01
0	1	1	846	.6318	.051
0	2	2	1355	.1.0119	.131
0	3	3	1359	.1.0149	.202
0	5	5	1371	.1.0232	.402
0	6	6	1362	.1.0172	.599

1	1	10 DEG R	10 DEG R	PML/P01	
2	2				
3	3				
4	5				
5	6				



PAGE = 1

10/20/71

AEDC (AHO, INC.) ARNOLD AFS, TENNESSEE
 VON KARMAN GAS DYNAMICS FACILITY
 50 INCH HYPERSONIC TUNNEL R
 Y1162

CH	POS	TAP	PR	PR/PO1	Y (IN)	PML/PR	ML	REL	TL/T- INF	UL/U- INF	RHOL/RHO- INF	MUL/MU- INF	
"	1	3	1	1.273E+00	9.922E-01	.014	3.235E-01	1.411	7.581E-05	9.869	.554	2.6951	
"	2	3	2	1.027E+01	1.401E+00	.066	2.291E-01	1.733	1.093E+06	8.620	.636	3.0858	
"	3	3	3	1.588E+01	2.167E+00	.012	1.482E-01	2.206	1.791E+06	.6993	.729	3.8036	
"	4	3	4	1.766E+01	2.409E+00	.00	1.333E-01	2.335	2.035E+06	.602	.750	4.0292	
"	5	3	5	1.846E+01	2.51HE+00	.00	2.16	2.515E-01	2.392	2.150E+06	.437	.759	4.1324
"	6	3	6	1.877E+01	2.560E+00	.00	2.58	1.254E-01	2.413	2.196E+06	.375	.762	4.1722
"	7	3	7	1.917E+01	2.561E+00	.00	2.413	1.253E-01	2.410	2.196E+06	.375	.762	4.1722
"	8	3	8	1.917E+01	2.561E+00	.00	2.413	1.253E-01	2.410	2.196E+06	.375	.762	4.1722
"	9	3	9	1.965E+01	2.544E+00	.00	2.365	1.259E-01	2.407	2.183E+06	.392	.761	4.1613
"	10	3	10	1.75AE+01	2.39HE+00	.00	2.415	1.262E-01	2.405	2.179E+06	.398	.761	4.1577
"	11	3	11	1.809F+01	2.444E+00	.00	2.499	1.33AE-01	2.329	2.023E+06	.619	.749	4.0187
"	12	3	12	1.822E+01	2.446E+00	.00	2.606	1.301E-01	2.366	2.098E+06	.510	.755	4.0859
"	13	3	13	1.813F+01	2.474E+00	.00	2.702	1.291E-01	2.374	2.114E+06	.487	.756	4.1001
"	14	3	14	1.795E+01	2.449E+00	.00	2.74	1.298E-01	2.368	2.102E+06	.504	.755	4.0894
"	15	3	15	1.785E+01	2.435E+00	.00	2.981	1.318E-01	2.349	2.079E+06	.539	.753	4.0681
"												3.0539	
"												3.0489	
"												3.0588	

CH	TC	TTR	TH/TG	Y (IN)	PML/P01
"	1	1	720	.5372	.051
"	2	2	1316	.9821	3.210E-01
"	3	3	1357	1.0127	.131
"	5	5	1367	1.0201	.202
"	6	6	1358	1.0134	.599

