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Summary Report

SHUTTLE OPERATIONS SIMULATION MODEL PROGRAMMERS'/USERS' MANUAL

June 1972

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SUMMARY REPORT
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SHUTTLE OPERATIONS SIMULATION MODEL
PROGRAMMERS'/USERS' MANUAL

By

D. G. Porter

June 1972

Prepared For

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MECHANICAL AND CREW SYSTEMS INTEGRATION DIVISION
ASTRONAUTICS LABORATORY
GEORGE C. MARSHALL SPACE FLIGHT CENTER

Contract No. NAS8-21804

**Details of illustrations in
this document may be better
studied on microfiche**

Prepared By

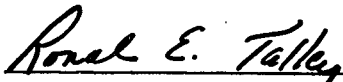
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ABSTRACT

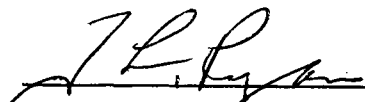
The prospective user of the Shuttle Operations Simulation (SOS) Model is given sufficient information by this document to enable him to perform simulation studies of the Space Shuttle launch-to-launch operations cycle. The procedures used for modifying the SOS Model to meet user requirements are described. The various control card sequences required to execute the SOS Model are given in the text of the report. The report is written for users with varying computer simulation experience.

A description of the components of the SOS Model is included that presents both an explanation of the logic involved in the simulation of the Shuttle operations cycle and a description of the routines used to support the actual simulation.

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DEFINITION OF SYMBOLS

CA	Cards added
CD	Cards deleted
CPU	Central Processing Unit
GPSS	General Purpose Systems Simulator
LUT	Launch Umbilical Tower
MSFC	Marshall Space Flight Center
SOS	Shuttle Operations Simulation

Control Card Sequence--A sequence of UNIVAC 1108 instructions that represents the information required to execute a computer program.

Insert Card--A specific card type used to indicate the addition to or the deletion of the statements contained in the computer program.

Location Number--The UNIVAC 1108 computer assigned number to each statement of the computer program.

Mnemonic Inserts--The GPSS II language commonly uses numeric block identification numbers. The language can accept alphanumerics as block numbers by using specific control card sequences. The alphanumeric block identification numbers are referred to as mnemonic inserts.

Model--A representation of a process or concept that permits the manipulation of its parameters as a means of determining how the concept or process behaves in various situations.

Operational Cycle--Those functions performed on each Shuttle element after its previous launch in preparation for its succeeding launch.

Simulation--Represents the behavior of a physical system by means of a computer and a program. The computer and program serves to demonstrate system behavior under the influence of stimuli and permits system performance analysis.

Subprogram--A routine designed to solve a particular problem that becomes applicable to the problem only when appropriate parametric values are supplied.

DEFINITION OF SYMBOLS (Concluded)

Supplemental Fortran Instructions--A collection of Fortran routines, both user written and GPSS II language supplied, that implement and support the simulation of the Shuttle turnaround cycle.

System Model Description--A sequence of GPSS II language instructions that, upon implementation, simulate the Space Shuttle launch-to-launch operations cycle.

Turnaround Cycle--Starting with a Shuttle vehicle at launch, represents those operations performed to prepare the vehicle to the same state of preparedness for its next launch.

SECTION I. INTRODUCTION

A. Purpose

A model has been developed that analyzes the major events of the Space Shuttle launch-to-launch operations cycle. The formulation of a model describing the Shuttle operations cycle was initiated by isolating the operational characteristics of the elements involved in the operations cycle. Next, the logical rules governing the interaction of those operational characteristics were identified and formulated into a model describing the entire Shuttle launch-to-launch operations cycle. To further analyze the system, a simulation technique was employed as an effective method of testing and evaluating the proposed real system. The simulation technique employed allows statistical estimates of the operational parameters in the real system. Using the results of the simulation technique only allows comparison of the alternatives to the real system, and does not necessarily generate an optimal solution. The resulting model has been named the Shuttle Operations Simulation (SOS) Model and, as developed, is activity oriented and designed to simulate the major events of the Space Shuttle launch-to-launch operations cycle. The SOS Model serves as a tool for conducting operations and logistical analysis relevant to the definition and design phase of the Space Shuttle program.

The Space Shuttle operational cycle is described by the General Purpose Systems Simulator (GPSS) II computer language which is supported by special purpose Fortran subprograms. As written, the SOS Model is capable of projecting real life activities and functions which will probably occur during the operational phase of the Space Shuttle program. From the results of parametric studies, real life projection and logistical and operational requirements can be determined for the Space Shuttle system.

This document is written to describe the use and operations of the SOS Model. Sufficient data and instructions are given to the potential user of the SOS Model to enable him to modify the model to meet his particular needs. A complete description of the components of the SOS Model is also given within the text of this document.

B. Scope

This report provides instructions for the use and operation of the SOS Model. Supplemental information is given to allow user modifications to the program enabling the user to meet his particular requirements. Section II of this document contains general information relating to the SOS Model operations and a description of the necessary input data required for implementing the SOS Model. Section III describes the two basic components of the SOS Model, the System Model Description, and the Supplemental Fortran Instructions. Included in Section IV are the control card sequences used to execute the SOS Model along with a description of each control card sequence. The Appendices contain data to support the sections of this document.

SECTION II. SOS MODEL OPERATING INSTRUCTIONS

A. General Information

The SOS Model is designed for execution by the Marshall Space Flight Center (MSFC) UNIVAC 1108 EXEC VIII Computer System and requires approximately 32,800 words of core storage. The SOS Model consists of two major components: the System Model Description and the Supplemental Fortran Instructions. The System Model Description component consists of a sequence of GPSS II language instructions that describe the launch-to-launch operation cycle of the Space Shuttle flight elements and support hardware. The Supplemental Fortran Instructions are a combination of user-written Fortran subprograms and the standard subprograms supplied as part of the GPSS II language. The GPSS II supplied subprograms are responsible for implementing the instructions contained in the System Model Description.

The SOS Model is maintained as a catalogued drum file by the UNIVAC 1108 computer located at MSFC. The catalogued file contains both the Supplemental Fortran Instructions and the System Model Description which defines a baseline configuration of the operational elements and their time distribution parameters. The System Model Description, as catalogued, defines a specific operational concept for the Space Shuttle launch-to-launch operations cycle. Using the basic configuration, a number of Shuttle flight elements and support facility arrangements are capable of being studied. Therefore, the user must decide the type of analysis to be performed and select the parameters to be varied.

B. Temporary System Model Modification Procedure

If the catalogued version of the System Model Description of the SOS Model is not entirely satisfactory to the user, temporary modifications to the System Model Description can be accomplished by utilizing the procedure described below. In making temporary model modifications, it will be useful to refer to Appendix A which contains a printout of the System Model Description. This printout is fully documented with each of the GPSS II language system variables used in the System Model Description defined prior to the first GENERATE block. Section IV shows the appropriate placement of the Model modification cards in the control card sequences required to execute the SOS Model.

Each statement of the System Model Description is computer numbered with the number positioned to the extreme left of each statement. Appendix A contains a computer printout of the System Model Description showing the GPSS II statements, block numbers, and computer assigned location numbers. To temporarily modify the System Model Description, such as to change the Booster and Orbiter storage capacities or to change the number of launch pads, the insert card method is used.

The insert card method requires an insert card which contains any of the following: the location number of the statement to be replaced (Example 1), the sequential location numbers of the statements to be modified (Example 2), or the location number of the statement which modifications are to follow (Example 3). All insert cards are punched beginning with a minus sign in column one. The inserted statements must conform with GPSS II block type formats (Reference 2). The following examples illustrate the use of the insert card method for implementing temporary model modifications. The format for these examples is: original GPSS II statements from the System Model Description, solid line, insert cards and new GPSS II statements. A brief discussion follows each example.

Example 1:

000051	2	CAPACITY	2
000052	3	CAPACITY	2
000053	4	CAPACITY	9

-51,51			
2	CAPACITY	1	
-53,53			
4	CAPACITY	7	

In Example 1 the storage capacities of both Booster maintenance and Booster storage, identified as location Nos. 51 and 53, respectively, were modified. The storage capacity of Booster maintenance was changed from two to one, and the Booster storage capacity was changed from nine to seven. Notice on each insert card that the location number is listed twice. This example illustrates a one for one replacement of the statement whose location was 53. It should be noted that all statements between the insert card and the succeeding insert card, if any, will replace the statement at location 53. Thus, it is possible to delete the statement at location 53 or to insert a finite number of statements in place of the statement at location 53.

Example 2:

000000	11	CAPACITY	2
000001	12	CAPACITY	2
000002	13	CAPACITY	9

-60,02			
11	CAPACITY	3	
12	CAPACITY	1	
13	CAPACITY	8	

Example 2 illustrates the modification procedure required when replacing all statements between and including two statement location numbers. The storage capacities of Orbiter maintenance, Orbiter test, and Orbiter storage were changed from two, two, and nine, respectively, to three, one and eight, respectively. The insert card contains the two numbers, inclusive, that identify the program segment to be modified. Using the above example and the discussion at the conclusion of Example 1, it is possible to delete all statements between and including two location numbers or to insert a finite number of statements in place of the statements between and including the indicated location numbers.

Example 3:

000000	31	CAPACITY	1
--------	----	----------	---

-80			
32	CAPACITY	3	
33	CAPACITY	4	
34	CAPACITY	3	

Example 3 illustrates how a number of statements can be inserted to follow a particular statement. It should be noted that only one location number is required on the insert card with this number preceded by a minus sign.

Example 4:

000489	70	QUEUE	21		ALL	71	72
--------	----	-------	----	--	-----	----	----

-489,489							
70	QUEUE	21		ALL	71	73	

Example 4 changes the System Model Description from the use of two launch pads to the use of three launch pads. By using the following QUEUE card, instead of the QUEUE card shown previously in this example, the System Model Description can be modified to limit the number of launch pads to one.

70	QUEUE	21		71
----	-------	----	--	----

It should be noted that the maximum number of available launch pads is three.

C. Temporary Supplemental Fortran Instructions Modification Procedure

Temporary modifications to the subprograms contained in the Supplemental Fortran Instructions can be accomplished by utilizing the insert card method outlined in the preceeding discussion. Appendix B contains a printout of the Supplemental Fortran Instructions. Like the System Model Description, each statement of the Supplemental Fortran Instructions has a computer assigned location number. These location numbers are used in conjunction with the insert card method previously outlined to perform the temporary modifications required by the user of the SOS Model.

Mention of the modification procedure to the Supplemental Fortran Instructions has been made only for completeness. Thus, no effort has been made to include examples of modifications. Modifications to the Supplemental Fortran Instructions should be attempted by only the most experienced user. Section IV shows the control card sequence required when modifications are made.

D. Input Data Description

The data input requirements for the execution of the SOS Model consist of the two card types described below.

Card type one, which is commonly referred to as the comment card, is used by the SOS Model to place a desired comment or heading on the output summary tables resulting from the execution of the SOS Model. All 80 columns of the card may be used to obtain the heading for the output summary tables.

Card type two, which is commonly referred to as the data card, defines the traffic density to be scheduled, the number of Booster and Orbiter elements to be used, the number of active Launch Umbilical Towers (LUT's), the level of significance for the statistics collected by the Confidence Interval Option (Reference 3), and the random number generator seeds for both the Traffic Model Simulator and the Shuttle Operations Simulator. The user has the option of choosing any of the 5 available traffic densities, numbered consecutively from 1 to 5 which represent 20, 35, 45, 55, and 75 launches per year, respectively. The required format for the type two data card is:

Column 1	Traffic Density Required
Columns 2 through 5 (right justified)	Number of Boosters
Columns 6 through 10 (right justified)	Number of Orbiters
Columns 11 through 15 (right justified)	Number of LUT's
Columns 16 through 20 (right justified)	Level of Significance for Confidence Interval Option
Columns 61 through 70	Random Number Seed for the Traffic Model Simulator
Columns 71 through 80	Random Number Seed for the Shuttle Operations Simulator

Sample data cards can be seen in Section IV which contains the control card sequences used to execute the SOS Model.

The Fortran format associated with data card type two in SUBROUTINE HELP is:

```
FORMAT (I1, I4, 3I5, 40X, 2I10) .
```

Following the second data card type, it is necessary to place a card which terminates the computer's execution of the current set of data. This is accomplished by placing (after the data card) a card that has the word END punched in columns seven through nine.

SECTION III. SOS MODEL DESCRIPTION

A. SOS Model Components

The SOS Model consists of two major components: the System Model Description and the Supplemental Fortran Instructions (Figure 1). The model, formed by analyzing the operational elements of the Space Shuttle turnaround cycle, was translated into the GPSS II language. The resulting set of instructions that describe the model are known as the System Model Description. When the System Model Description is implemented, the launch-to-launch operations cycle of the Space Shuttle vehicle and support hardware is simulated. The Supplemental Fortran Instructions of the SOS Model contain the routines which implement the instructions contained in the System Model Description. The routines contained in the Supplemental Fortran Instructions are mostly user written Fortran routines. However, there are several routines supplied by the GPSS II language that are included in the Supplemental Fortran Instructions.

Reference 1 describes the SOS Model. Sections III B and III C give more detailed data relating to the two major components of the SOS Model.

B. SOS Model System Description

The System Model Description component of the SOS Model is a sequence of GPSS II language instructions that allows the simulation of the operational facts concerning individual Shuttle processing facilities, LUT's, Orbiters, and Boosters. The necessary instructions are included to simulate the operations involved in the complete launch-to-launch cycle of the Space Shuttle. The operations in the launch-to-launch cycle are expected to have time distributions associated with their performance. In order to reflect this, frequency distribution functions have been estimated that approximate the expected real life time distributions for the performance of each operation. The estimated frequency distribution functions are used in conjunction with the Monte Carlo technique to provide a stochastic simulation of the launch-to-launch operations for the Space Shuttle.

The following is a brief description of the overall logic in the System Model Description. Figure 2 is a detailed flowchart depicting

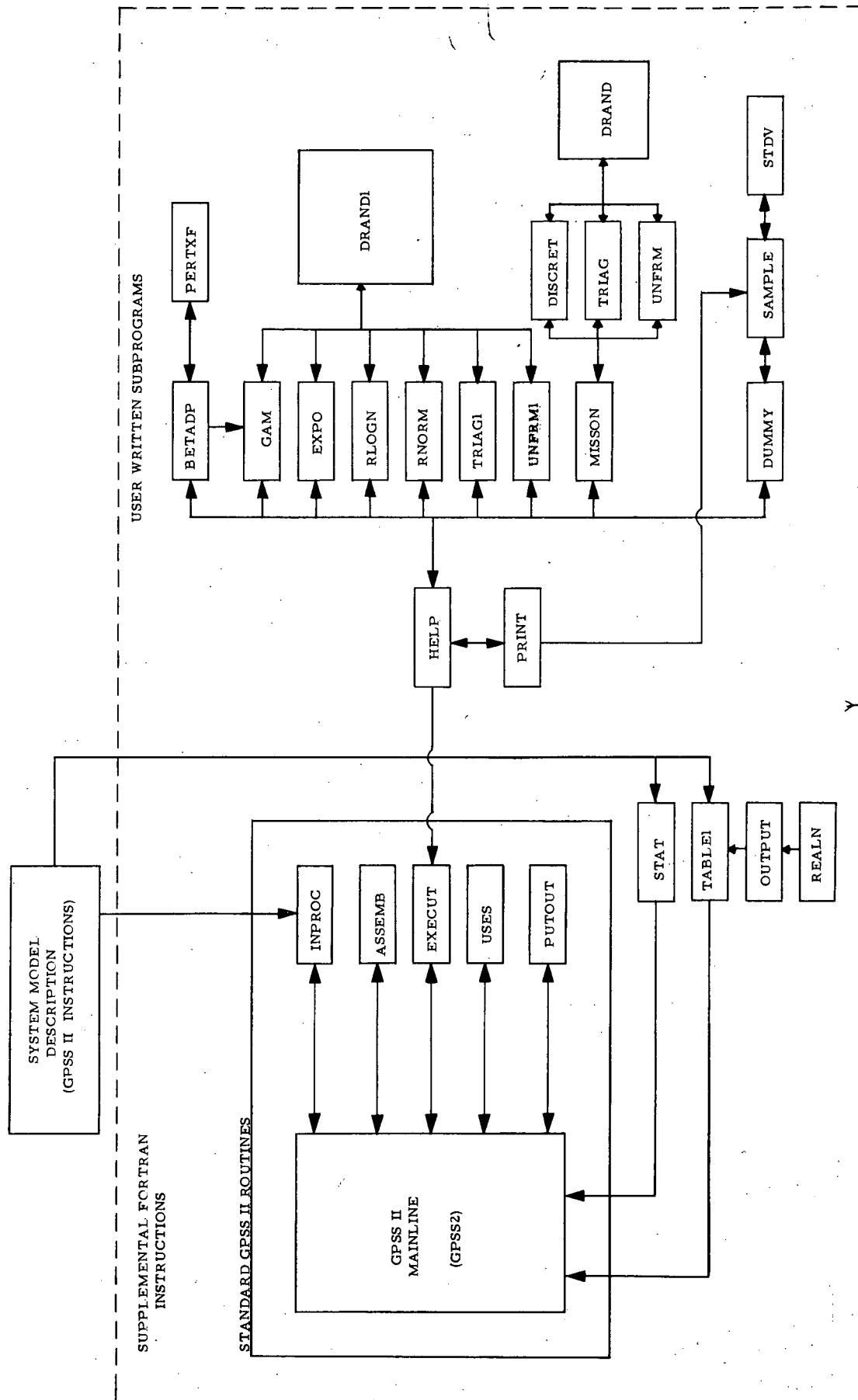


FIGURE 1. SOS MODEL CONFIGURATION



the functional logic of the Shuttle turnaround cycle. In the ensuing discussion, references are made to GPSS II system variables through the use of parentheses. A brief definition of the commonly used system variables is contained in Table I.

Using a GPSS II HELP block and Fortran SUBROUTINE HELP, a data card is read which contains the number of Boosters, Orbiters, and LUT's. Each of these values is placed in a separate storage location (SAVEX) for use during the simulation.

The simulation proceeds by using a HELP block to determine both a realistic first launch date and the parameters associated with the mission scheduled for that date. The HELP block, through Fortran SUBROUTINE MISSION, also establishes the time between launches. Throughout the simulation, the launch intervals and the parameters for both the current and the succeeding missions are always known.

The earliest feasible storage exit time prior to launch is the maximum processing time required to prepare the vehicle for launch. The request for vehicle elements for the current mission is not made until this time. The ready storage time prior to storage exit is determined by subtracting both the current simulated clock time (C1) and the maximum time required to process the vehicle from mate and hookup initiation to launch (V5) from the launch date of the current mission (X21). At the maximum processing time prior to the scheduled mission, a request is made to determine if there is a Booster, Orbiter, and LUT available in their respective storages. If all three elements are available, their storage departure time is recorded and they are placed in a queue for the mate and hookup operation (Q20).

The time a request may be delayed (V11) because of vehicle element unavailability is the difference between the maximum and minimum time to process the vehicle from mate and hookup initiation to launch. If this time is exceeded, the mission is cancelled and a record of the unavailable elements is kept. When all vehicle elements are available and the maximum time delay has not occurred, the pad availability is checked. The vehicle elements are removed from storage, sent to the mate and hookup waiting line (Q2), and this information is recorded. If a pad will not be available, the mission is cancelled and the pad unavailability is recorded.

When the vehicle elements complete mate and hookup (S20), the vehicle goes to the launch pad waiting line (Q21) where it is

TABLE I. DEFINITION OF COMMONLY USED SYSTEM VARIABLES*

Scope	Name	Meaning
Equipment	Fn	Facility status, facility n
	Sn	Storage occupancy, storage n
Queue	Qn	Queue length, queue n
System	Vn	Defined variable, numbered n
	Xn	Stored value, cell n
	Kn	Constant n

*A more complete description of the available GPSS II system variables can be found in Reference 2.

assigned to the first available pad after the pads have been checked in sequential order. (After a vehicle has been assigned a pad, the pad is unavailable until either the vehicle has been launched and the pad refurbished or until the vehicle is removed from the pad.)

Once a pad has been assigned to a vehicle and a LUT, they are transferred to the pad from the launch pad waiting line (Q21). The vehicle is installed on the pad and precountdown preparations are performed (S22). The countdown time (V6) is compared to the time left to launch (V2) to determine if sufficient time remains to complete countdown. If excess time exists to launch, a built-in variable length hold (V3) is utilized. When insufficient time remains to meet the first launch window, the second launch window is tried.

Should a failure occur while the vehicle is in countdown, a check is made to determine if the failure can be corrected on the pad. If the failure cannot be corrected on the pad, the vehicle and LUT are removed from the pad (S25) and the pad is released for the use of another mission. The Orbiter is separated from the Booster and LUT (S30) and sent to maintenance (S11). The Booster is separated from the LUT (S19) and they are sent to their respective maintenances (S2, S18). If no failure occurs, the vehicle is launched. The LUT is removed from the pad (S17) and sent to maintenance (S18) and the launch pad is refurbished (S27) and released for future launches. The lift-off time, time between lift-off and separation, Booster flight time, and Orbiter flight time are recorded.

After the Booster flies back and lands, it enters the waiting line for Booster safing (Q1) and enters the safing facility (S1) when it becomes available. Upon completion of safing, the Booster enters the maintenance waiting line (Q2) and maintenance is performed. When maintenance is completed, the Booster enters storage and becomes available for future launch requests.

After the Orbiter has performed its mission, flown back to landing site and landed, it enters the waiting line for Orbiter safing (Q10). When the safing facility (S10) becomes available, the Orbiter is safed and transported to the Orbiter maintenance waiting line (Q11). After maintenance is completed on the Orbiter, it is transported to storage for future use.

When the above Shuttle operations cycle logic is simulated, three computer outputs that describe the simulation results are available to the user of the SOS Model. Sample computer outputs resulting from executing the SOS Model can be found in Appendices C, D, and E.

C. SOS Model Supplemental Fortran Instructions

The Supplemental Fortran Instructions component of the SOS Model is composed of user written Fortran routines and the GPSS II supplied routines. The instructions necessary to simulate the System Model Description of the SOS Model are supplied by the Supplemental Fortran Instructions. Figure 1 denotes the manner in which the System Model Description interfaces with the Supplemental Fortran Instructions.

The GPSS2 routine, as supplied by the GPSS II language, is the nucleus of the SOS Model and, as such, actually performs the operations required to execute the System Model Description which in turn simulates launch-to-launch operations cycle of the Space Shuttle. All of the GPSS II supplied routines are in binary format and cannot be changed, with the exception of the GPSS2 routine. Consequently, the GPSS2 routine is the only GPSS II supplied routine discussed within this document. At the discretion of the user, this routine may be altered through the use of Fortran parameter statements to increase or decrease the size limitations of the System Model Description. However, this should be attempted by only the most experienced of users.

Upon request, the GPSS2 routine calls either of two user written Fortran subroutines: TABLE1 and STAT. Both subroutines are used to read the System Model Description of the SOS Model as data. Both subroutines are described below.

TABLE1 is used to output a table of time distribution parameters for each of the operations simulated by the System Model Description. The output of the time distribution parameters is accomplished by reading each statement of the System Model Description until a comment card is located that has the key symbol IDST in columns 68 through 71, with columns 67 and 72 containing a blank. The program continues reading cards until a HELP card is found which is used to supply the desired information for the time distribution parameters. The program prints the title of the activity found on the comment card and the time distribution parameters found on the HELP card. Appendix C contains an example listing of the TABLE1 output. It should be noted that TABLE1 and its support routines of OUTPUT and REALN are not active during the dynamic simulation of the Space Shuttle operations cycle. Section IV gives the control card sequences required to obtain the TABLE1 output.

STAT is used to read and store headings to be in the output of confidence interval statistics. A detailed description of the use of the Confidence Interval Option is contained in Reference 3. Each statement of the System Model Description is read until a statement with the key symbol CONF in columns 67 through 70 is found. At such time, the heading contained on that statement is stored to be used during the output of the confidence interval statistics. Appendix E contains an example output from the Confidence Interval Options. STAT, like TABLE1, is not active during the dynamic simulation of the operations cycle of the Space Shuttle. A complete description of the use of SUBROUTINE STAT and of the Confidence Interval Option can be found in Reference 3.

The primary user written routine of the Supplemental Fortran Instructions is the HELP routine. The HELP routine enables the GPSS II user to perform operations that are not easily obtainable from using the GPSS II block types. Upon completion of the operations in the HELP routine, the calculated values are transmitted back to the System Model through a specified SAVEX (system variable Xn), which is a core storage location. The user employing the GPSS II HELP block must supply a Fortran subroutine called HELP which has five or less fixed point variables as arguments. These arguments are specified on the GPSS II HELP card in the X, Y, Z, Mean, and Mod field locations.

The launch requirements for the Space Shuttle and the Orbiter on-orbit mission requirements are supplied by SUBROUTINE MISSION which is supported by subprograms UNFRM, DICRET, TRIAG, and DRAND. To gain access to SUBROUTINE MISSION, one must use the HELP block in the GPSS II model as illustrated in the following example:

```
* CALL HELP TO GENERATE ATTRIBUTES OF NEXT LAUNCH
24  HELP          K1      X11    X12    BOTH  10    25    X13    X10
```

Upon entering the HELP block, the program calls for SUBROUTINE HELP with its five fixed point variables. The proper form of SUBROUTINE HELP is shown below:

```
SUBROUTINE HELP(K1,K2,K3,K4,K5)
```

The first variable in the call statement, K1, is used by a Fortran computed GO TO statement within SUBROUTINE HELP to determine which Fortran user written routine is to be used to assist the operations of the GPSS II block types. In the above example, the value of K1 is used to call SUBROUTINE MISSION from SUBROUTINE HELP.

SUBROUTINE MISSION calculates the length of the next Orbiter mission, determines the mission type, computes the time until the next launch, and the time between launch opportunities. These values are calculated as different variables in SUBROUTINE MISSION and reset equal to K2, K3, K4, and K5 by SUBROUTINE HELP. The calculated values are then stored in the SAVEX locations of X11, X12, X13, and X15 for use by the GPSS II program.

SECTION IV. UNIVAC 1108 CONTROL CARD SEQUENCES USED IN THE EXECUTION OF THE SOS MODEL

A. General Discussion

Execution of the SOS Model can be accomplished by choosing the appropriate control card sequence of those described in this section. Multiple executions of the SOS Model can be accomplished by using the Systems Model Description as catalogued on the UNIVAC 1108 drum files. Two executions, both with and without the Confidence Interval Option and each simulating 500 Shuttle launch requests, can be obtained from one computer run with a resulting Central Processing Unit (CPU) time of approximately 3 minutes.

In selecting the appropriate control card sequence, it is assumed that the user has acquainted himself with the material and techniques presented in this document, and that he has defined his objectives sufficiently to enable an intelligent decision in the choice of control card sequences. The control card deck setups which follow are used in conjunction with the execution of the SOS Model. No effort has been made to enumerate all the possible control card sequences. Instead, only the most commonly used sequences are presented. A description of each individual control card is provided in Appendix F.

B. Multiple Executions of the SOS Model

The basic control card sequence used for executing the SOS Model, when no modification or additions are made to either the System Model Description or the Supplemental Fortran Instructions, is shown in Figure 3. The sequence (Figure 3) includes two of the options available to the user of the SOS Model. Both options are for the convenience of the user and the inclusion or deletion of either option has no effect on the results of the simulation.

The @SETC 4 control card placed before the @XQT control card generates the TABLE1 output as described previously in this document. In order to produce the output table, it was necessary to place the System Model Description in the execution run stream. This was accomplished by the addition of the one extra @ADD MODEL control card placed immediately after the first @XQT control card.


```

10 RUN, / / T SAMPLE, 999999, PORTERBIN225, 3, 150
10 ASG, 1 GPSSE8, F2
10 COPY, RSA MASTER*GPSSE8., GPSSE8.
10 ASG, 1 MODEL, F2
10 DATA FILE2, MODEL
10 END
10 SETC 4
10 XGT GPSSE8.MAPGPS
10 ADD MODEL.
10 ADD MODEL.
*****COMMENT CARD*****
*****DATA CARD*****
      END
10 SETC 3
10 XGT GPSSE8.MAPGPS
10 ADD MODEL.
*****COMMENT CARD*****
*****DATA CARD*****
      END
10 FIN
10 FIN

```

FIGURE 3. MULTIPLE EXECUTION OF THE SOS MODEL
WITHOUT MODIFICATIONS

The @SETC 3 control card placed before the second @XQT control card eliminates the printout of the model listing but allows the output of the summary results and the standard GPSS II output.

C. Modifications to the System Model Description

If the catalogued version of the System Model Description is not completely satisfactory to the user of the SOS Model, temporary model changes can be made through the use of the control card sequence shown in Figure 4.

The control card sequence shown in Figure 4 is used for multiple executions of the SOS Model. The sequence also encompasses the use of the TABLE1 output and the use of the @SETC 3 command. If only one execution is desired, the two @FIN cards should follow the first END card.

D. Modifications to the Supplemental Fortran Instructions

The user written routines of the Supplemental Fortran Instructions can be added to or modified in order to meet the user's requirements by utilizing the control card sequence shown in Figure 5. The use of this control card sequence enables the user to test both a modified user written routine and a newly developed user written routine by utilizing the catalogued drum files. The control card sequence (Figure 5) is an example of only one execution of the SOS Model; but, multiple executions are possible by utilizing the information previously supplied in this section. The procedure for modifying the user written routines entails the use of the insert card method described previously in conjunction with modifications to the System Model Description. More detailed information concerning the user written routines can be found in Appendix B.

E. Creation of an Update Tape

The control card sequence (Figure 6) is used to create a tape that includes all updates to both the Supplemental Fortran Instructions and the System Model Description. The created tape, which includes all updates, can then be used to update the catalogued drum file of the SOS Model. Great care should be taken when creating a tape to be used for updating the SOS Model. To ensure the proper operation of the updated tape, Figure 7 gives the control card sequence used to test the newly created tape.

```

10 RUN, /Z/T SAMPLE, 999999, PORTERBIN225, 3, 150
10 ASG, T GPSSE8, F2
10 ASG, T MODEL, F2
10 COPY, RSA MASTER*GPSSE8., GPSSE8.
10 DATA FILE2, MODEL
10 ****MODEL CHANGE CARDS*****
10 ****(IF REQUIRED)*****
10 END
10 SETC 4
10 XQT GPSSE8.MAPGPS
10 ADD MODEL.
10 ADD MODEL.
10 *****COMMENT CARD*****
10 *****DATA CARD*****
10      END
10 SETC 3
10 XQT GPSSE8.MAPGPS
10 ADD MODEL.
10 *****COMMENT CARD*****
10 *****DATA CARD*****
10      END
10 FIN
10 FIN

```

FIGURE 4. MULTIPLE EXECUTION OF THE SOS MODEL WITH
SYSTEM MODEL DESCRIPTION MODIFICATIONS

```

10RUN,ZZT SAMPLE,999999,PORTERBIN25,3,150
10ASG,T GPSSE8,F2
10ASG,T MODEL,F2
10COPY,RSA MASTER*GPSSE8.,GPSSE8.
*** ANY SUPPORTING FORTRAN SUBPROGRAMS WHICH ARE
*** BEING MODIFIED SHOULD BE INSERTED AS FOLLOWS.
*** (REPLACE 'NAME' WITH THE APPROPRIATE SUBPROGRAM NAME)
10FOR,FUS GPSSE8.NAME,GPSSE8.NAME
*** INSERT THE CHANGES TO SUBPROGRAM 'NAME'.
***
*** ANY SUPPORTING FORTRAN SUBPROGRAMS WHICH ARE
*** BEING ADDED SHOULD BE INSERTED AS FOLLOWS.
*** (REPLACE 'NAME' WITH THE APPROPRIATE SUBPRGRAM NAME)
10FOR,IS GPSSE8.NAME,GPSSE8.NAME
*** INSERT THE ADDED SUBPROGRAM 'NAME'.
10HUG,P (IN THESE SPACES,ANY DESIRED COMMENT)
10PREP GPSSE8.
10MAP,NX GPSSE8.MAP,GPSSE8.MAPGPS
10DATA FILE2,MODEL
*****MODEL CHANGE CARDS*****
***** (IF REQUIRED)*****
10END
10XQT GPSSE8.MAPGPS
10ADD MODEL.
*****COMMENT CARD*****
*****DATA CARD*****
      END
10FIN
10FIN

```

FIGURE 5. EXECUTION OF THE SOS MODEL WITH SUPPLEMENTAL FORTRAN INSTRUCTIONS MODIFICATIONS

```

10 RUN, /Z/ T SAMPLE, 999999, PORTERBIN225, 3, 200/5000
10 ASG, T TAPENAME, 1, SAVE02
10 ASG, T GPSSE8, F2
10 ASG, T DUMMYP, F2
10 ASG, T MODEL, F2
10 COPY, RSA MASTER*GPSSE8., GPSSE8.
10 HDG, P <<<PROGRAM FILE UPDATE NUMBER AAA>>>
***INSERT ALL SUBPROGRAM UPDATES AND INSERTIONS HERE***
10 PREF GPSSE8.
10 MAP, XS GPSSE8.MAP, GPSSE8.MAPGPS
10 HDG, P <<<SOS MODEL UPDATE MSFC VERSION III-BBB>>>
10 DATA, L FILE2, MODEL
-1, 1
      ASSEMBLER
      JOB
***INSERT ALL MODEL MODIFICATIONS HERE***
-XXX, XXX
      END
10 END
10 SETC 2
10 BRKPT PUNCH$/DUMMYP
10 XQT GPSSE8.MAPGPS
10 ADD MODEL.
10 BRKPT PUNCH$
10 DATA, L DUMMYP, MODEL
-YYY, YYY
      START          501
10 END
10 REWIND TAPENAME
10 CPOUT GPSSE8, TAPENAME
10 COPY, GM MODEL, TAPENAME
10 ERS GPSSE8.
10 REWIND TAPENAME
10 COPIN TAPENAME, IPF$.
10 FREE TAPENAME.
10 HDG, P <REVISION NUMBER CCC TO THE SYSTEM FILE>
10 PRT, T
10 XQT, LA SYS$*MSFC$.LISTIT
10 FIN

```

FIGURE 6. CREATION OF A TAPE FOR UPDATING THE SOS MODEL

A few words of explanation are needed before the user attempts to create a tape containing the inserts and additions to either of the major components of the SOS Model. In the control card sequence (Figure 6), it should be noted that there are two insert cards, one with a location number of XXX and one with a location number of YYY. Before attempting to create the tape, location number XXX needs to be replaced by the location number of the GPSS II START card. At this time in the control card sequence, the System Model Description is assembled with all the statements reassigned new location numbers. The START card has been temporarily replaced by an END card which prohibits the execution of the model and prevents an error occurring during the assembly process. Location number YYY needs to be the new location number of the inserted END card and is determined by knowing the total number of new cards added (CA) and the number of cards deleted (CD) from the current version of the model. The new insert number of the END statement is found by:

$$YYY = XXX + CA - CD$$

At this time in the control card sequence, the START card permanently replaces the END card prior to storing the System Model Description on tape.

The sample control card sequence (Figure 7) is an example of one execution of the SOS Model. Multiple executions are possible by following the procedures outlined previously in this section. The numbers XXXXX on the @ASG, T TAPENAME card should be replaced by the number of the tape to be tested.

F. Mnemonic Inserts to the System Model Description

The control card sequence (Figure 8) is used when the modifications to the System Model Description contain mnemonic location references in the GPSS block types. Mnemonic references are helpful during the development of major modifications to the SOS Model when block location numbers are in a constant state of change. The presence of an insert card containing location number XXX should be noted in Figure 8. The number XXX should be changed to the location number of the START card before an attempt is made to execute this control card sequence. This will place an end card immediately after the START card as is required by mnemonic decks.

```

        RUN, //T SAMPLE, 999999, PORTERBIN225, 3, 150
        ASG, I TAPENAME, T, XXXXX
        ASG, T GPSSE8, F2
        ASG, I FILE2, F2
        ASG, I MODEL, F2
        REWIND TAPENAME
        COPYIN TAPENAME, GPSSE8.
        COPY, G TAPENAME, FILE2
        FREE TAPENAME
        DATA FILE2, MODEL
        END
        XQT GPSSE8.MAPGPS
        ADD MODEL.
        *****COMMENT CARD*****
        *****DATA CARD*****
        END
        FIN
        FIN

```

FIGURE 7. TEST OF AN UPDATE TAPE

```

        RUN, //T SAMPLE, 999999, PORTERBIN225, 3, 150
        ASG, T GPSSE8, F2
        COPY, RSA MASTER*GPSSE8., GPSSE8.
        ASG, T MODEL, F2
        *** INSERT ALL SUBPROGRAM UPDATES OR INSERTIONS HERE ***
        DATA FILE2, MODEL
        -1, 1
        ASSEMBLER
        JOB
        ***INSERT ALL MODEL MODIFICATIONS HERE***
        -XXX
        END
        END
        XQT GPSSE8.MAPGPS
        ADD MODEL.
        *****COMMENT CARD*****
        *****DATA CARD*****
        FIN
        FIN

```

FIGURE 8. EXECUTION OF THE SOS MODEL WITH MNEMONIC INSERTS TO THE SYSTEM MODEL DESCRIPTION

G. Example Production Deck and Explanation

Figure 9 is an example of an SOS Model production deck with temporary System Model Description modifications. The System Model portion was modified to change the statements at locations 51, 489, and 513. The modification at location 51 altered the capacity of the Booster maintenance facility and the modification at location 489 changed the number of available launch pads to 3. The modification at location 513 resulted in a mean time for countdown preparations of 2.0 days being associated with block number 80.

Two executions of the SOS Model are required to complete the case study. The @SETC 4 command is used to obtain a listing of the time distribution parameters contained in the Systems Model Description. The @SETC 3 command is used to prevent a listing of the System Model Description for the last execution of the program. An example of the TABLE1 output from the above run is found in Appendix C.

The first requested execution of the SOS Model uses traffic density four with the number of LUT's and launch pads fixed at three and three, respectively. The number of active Boosters and Orbiters for the first study are four and five, respectively. The random number seeds for the entire case study are constant. The random number seed for the Traffic Model Simulator is 3154267131 and the random number seed for the Shuttle Operations Simulator is 4339968911. (Note: Any ten digit odd number with the last digit not ending in five may serve as a random number seed.) The second execution uses the same study conditions as the first execution with the exception of the number of Orbiters being changed to four. Sample output of the summary table and the GPSS II output resulting from the above case study can be found in Appendix D.


```

16RUN, //P SAMPLE, 999999, POKIEBIN25, 3, 150
16ASG, I GPSSE8, F2
16ASG, I MODEL, F2
16COPY, KSA MASTER*GPSSE8., GPSSE8.
16DATA FILE2, MODEL
-51, 51
2 CAPACITY 1
-489, 489
70 QUEUE 21 ALL 71 73
-513, 513
79 HELP K3 K1500 K2000 80 K2500 X10
16END
16SETC 4
16XGT GPSSE8.MAPGPS
16AUD MODEL.
16AUD MODEL.
DETERMINATION OF FLEET SIZE REQUIREMENTS
4 4 5 3
END
16SETC 3
16XGT GPSSE8.MAPGPS
16AUD MODEL.
DETERMINATION OF FLEET SIZE REQUIREMENTS
4 4 4 3
END
16FIN
16FIN
31542671314339968911

```

FIGURE 9. EXAMPLE PRODUCTION DECK FOR THE EXECUTION OF THE SOS MODEL

REFERENCES

1. Shuttle Operations Simulation Model Description. S&E-012-001-2H, Rev. A. Marshall Space Flight Center, Huntsville, Alabama, February 15, 1971.
2. Univac General Purpose Systems Simulator II. UP-4129 Reference Manual.
3. Ghiglieri, F.J. and Porter, D.G.: Requirements for the Integration of Confidence Interval Logic into the Shuttle Operations Simulation Model. Interim Report ASD-ASTN-1485, Teledyne Brown Engineering Company, Huntsville, Alabama, March 1972.

APPENDIX A. SYSTEM MODEL DESCRIPTION LISTING

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<<<SOS MODEL UPDATE MSFC VERSION III-D>>>

000001	JOB		IIIA 20
000002	*		IIIA 30
000003	*		IIID 10
000004	*SHUTTLE OPERATIONS SIMULATION MODEL MSFC VERSION III-D MAR. 31-1972		IIIA 50
000005	*		IIIA 60
000006	*		IIIA 70
000007	*		IIIA 80
000008	*		IIIA 90
000009	*		IIIA 100
000010	*	ALL TIMES EXPRESSED	IIIA 110
000011	*	IN THIS PROGRAM ARE IN	IIIB 10
000012	*	TERMS OF MILLI WORKING DAYS (1/1000 OF A WORKING DAY)	IIIA 120
000013	*	(WORKING DAY = 16 HOURS)	IIIA 130
000014	*		IIIA 140
000015	*		IIIA 150
000016	*	PERFORMANCE TIME DISTRIBUTION CODES	IIIA 160
000017	*		IIIA 170
000018	*	K3 TRIANGULAR	IIIA 180
000019	*	K4 NORMAL	IIIA 190
000020	*	K8 EXPONENTIAL	IIIA 200
000021	*	K9 LOGNORMAL	IIIC 11
000022	*	K11 CONSTANT	IIIC 12
000023	*	K12 UNIFORM	IIIA 220
000024	*	K13 BETA	IIIA 230
000025	*		IIIA 240
000026	*		IIIA 250
000027	*		IIIB 20
000028	*		IIIB 30
000029	*		IIIB 40
000030	*		IIIB 50
000031	*		IIIB 60
000032	*	BLOCKS - 400	IIIB 70
000033	*	FACILITIES - 15	IIIB 80
000034	*	STORAGES - 50	IIIB 90
000035	*	QUEUES - 50	IIIB 100
000036	*	LOGIC SWITCHES - 25	IIIB 110
000037	*	SAVEX LOCATIONS - 50	IIIB 120
000038	*	FUNCTIONS - 10	IIID 20
000039	*	TABLES AND QTABLES - 30 (COMBINED TOTAL)	IIIB 140
000040	*	VARIABLE STATEMENTS - 30	IIIB 150
000041	*	PARAMETERS - 20	IIIB 160
000042	*	TRANSACTION NUMBERS - 100	IIIB 170
000043	*		IIIB 180
000044	*		IIIB 190
000045	*		IIIA 260
000046	*		IIIA 270
000047	*		IIIA 280
000048	*		IIIA 290
000049	*		IIIA 300
000050	1	CAPACITY	IIIA 310
000051	2	CAPACITY	IIIA 320
000052	3	CAPACITY	IIIA 330
000053	4	CAPACITY	
000054	5	CAPACITY	

000055	*6	IIIA 340
000056	*7	IIIA 350
000057	*8	IIIA 360
000058	*9	IIIA 370
000059	10	IIIA 380
000060	11	IIIA 390
000061	12	IIIA 400
000062	13	IIIA 410
000063	14	IIIA 420
000064	*15	IIIA 430
000065	16	IIIA 440
000066	17	IIIA 450
000067	18	IIIA 460
000068	19	IIIA 470
000069	20	IIIA 480
000070	*21	IIIA 490
000071	22	IIIA 500
000072	23	IIIA 510
000073	*24	IIIA 520
000074	25	IIIA 530
000075	*26	IIIA 540
000076	27	IIIA 550
000077	28	IIIA 560
000078	29	IIIA 570
000079	30	IIIA 580
000080	31	IIIA 590
000081	*	IIIA 600
000082	*	IIIA 610
000083	*	IIIA 620
000084	*	IIIA 630
000085	*	IIIA 640
000086	*	IIIB 200
000087	*	IIIA 660
000088	*	IIIB 210
000089	*	IIIB 220
000090	*	IIIB 230
000091	*	IIIB 240
000092	* 1 - BOOSTER SAFING (Q)	IIIB 250
000093	* 2 - BOOSTER MAINTENANCE (Q)	IIIB 260
000094	* 3 - TRANSPORT BOOSTER TO SYSTEMS TEST, TEST, TRANSPORT TO STORAGE (Q)	IIIB 270
000095	* 4 - BOOSTER STORAGE	IIIA 700
000096	* 5	IIIA 710
000097	* 6	IIIA 720
000098	* 7	IIIA 730
000099	* 8	IIIA 740
000100	* 9	IIIA 750
000101	* 10 - ORBITER SAFING (Q)	IIIB 280
000102	* 11 - ORBITER MAINTENANCE (Q)	IIIB 290
000103	* 12 - TRANSPORT ORBITER TO SYSTEMS TEST, TEST, TRANSPORT TO STORAGE (Q)	IIIB 300
000104	* 13 - ORBITER STORAGE	IIIA 790
000105	* 14 - ORBITER BOOST TO ORBIT, PERFORM MISSION, RETURN	IIIA 800
000106	* 15	IIIA 810
000107	* 16 - LAUNCH UMBILICAL TOWER (LUTS) STORAGE	IIIB 310
000108	* 17 - REMOVE POST LAUNCH LUT FROM PAD	IIIA 830
000109	* 18 - TRANSPORT LUT TO MAINT, REFURBISH/TEST, TRANSPORT TO STORAGE (Q)	IIIB 320
000110	* 19 - SEPARATE BOOSTER FROM LUT, TRANSPORT BOOSTER TO MAINTENANCE	IIIA 850

STORAGE AND QUEUE DEFINITIONS

NOTE - THE DEFINITIONS WHICH ARE FOLLOWED BY THE NOTATION (Q) ALSO REFER TO A QUEUE WHICH IMMEDIATELY PRECEEDS THE STORAGE. THE DEFINITIONS WHICH ARE NOT FOLLOWED BY THE NOTATION (Q) HAVE NO QUEUE ASSOCIATED WITH THEM.

- * 1 - BOOSTER SAFING (Q)
- * 2 - BOOSTER MAINTENANCE (Q)
- * 3 - TRANSPORT BOOSTER TO SYSTEMS TEST, TEST, TRANSPORT TO STORAGE (Q)
- * 4 - BOOSTER STORAGE
- * 5
- * 6
- * 7
- * 8
- * 9
- * 10 - ORBITER SAFING (Q)
- * 11 - ORBITER MAINTENANCE (Q)
- * 12 - TRANSPORT ORBITER TO SYSTEMS TEST, TEST, TRANSPORT TO STORAGE (Q)
- * 13 - ORBITER STORAGE
- * 14 - ORBITER BOOST TO ORBIT, PERFORM MISSION, RETURN
- * 15
- * 16 - LAUNCH UMBILICAL TOWER (LUTS) STORAGE
- * 17 - REMOVE POST LAUNCH LUT FROM PAD
- * 18 - TRANSPORT LUT TO MAINT, REFURBISH/TEST, TRANSPORT TO STORAGE (Q)
- * 19 - SEPARATE BOOSTER FROM LUT, TRANSPORT BOOSTER TO MAINTENANCE

000111	* 20 - MATE AND HOOKUP BOOSTER, ORBITER AND LUT (Q)	IIIB 330
000112	* 21 - QUEUE IMMEDIATELY PRECEEDING THE PAD FACILITIES (NOTE- THERE IS	IIIB 340
000113	* NO ASSOCIATED STORAGE)	IIIB 350
000114	* 22 - COUNTDOWN PREPARATION	IIIA 880
000115	* 23 - COUNTDOWN	IIIA 890
000116	* 24	IIIA 900
000117	* 25 - REMOVE BOOSTER, ORBITER AND LUT FROM PAD	IIIA 910
000118	* 26	IIIA 920
000119	* 27 - REFURBISH LAUNCH PAD	IIIA 930
000120	* 28 - LIFTOFF THROUGH SEPARATION	IIIA 940
000121	* 29 - READY STORAGE FOR NEXT BOOSTER, ORBITER AND LUT (Q)	IIIB 360
000122	(IMMEDIATELY PRIOR TO THE BEGINNING OF THE MATE OPERATION)	IIIA 960
000123	* 30 - TRANSPORT VEHICLE, SEPARATE ORBITER, TRANSPORT ORBITER TO MAINT.	IIIA 970
000124	* 31 - RESTRICTS THE REQUEST FOR VEHICLE ELEMENTS TO ONE MISSION	IIIB 370
000125	AT A TIME. SUCCEEDING MISSIONS MUST WAIT UNTIL THE MISSION	IIIB 380
000126	IMMEDIATELY IN FRONT OF THEM HAS EITHER SUCCESSFULLY MET ITS	IIIB 390
000127	REQUEST OR HAS BEEN CANCELLED DUE TO AN EXCESSIVE WAIT FOR	IIIB 400
000128	VEHICLE ELEMENTS. (Q)	IIIB 410
000129		IIIA 990
000130		IIIA1000
000131		IIIA1010
000132		IIIA1020
000133		IIIA1030
000134		IIIA1040
000135		IIIA1050
000136	* 1 - IS A CONTINUOUS LINEAR FUNCTION WITH A SLOPE OF 1. ITS END	IIIC 20
000137	POINTS ARE LOCATED AT (Q,0) AND (999999,999999). IT IS USED	IIIB 420
000138	TO CONVERT THE INPUT SYSTEM VARIABLE SAVEX 10 (X10) INTO	IIIB 430
000139	FUNCTION 1 (FN1), WHICH CAN BE USED AS A MODIFIER IN AN ADVANCE	IIIB 440
000140	BLOCK. THE FUNCTION IS DESCRIBED SUCH THAT THE CONVERSION	IIIB 450
000141	IS ONE TO ONE (FN1=X10).	IIIB 460
000142		IIIB 470
000143	1 FUNCTION X10 C2	IIIA1080
000144	0 0 99999999999	IIIA1090
000145		IIIA1100
000146		IIIA1110
000147		IIIA1120
000148		IIIA1130
000149		IIIA1140
000150		IIIA1150
000151		IIIA1160
000152	NOTE A - UNLESS OTHERWISE STATED THE FOLLOWING DEFINITIONS	IIIB 480
000153	REPRESENT THE TOTAL COUNT OF THE TRANSACTIONS	IIIB 490
000154	ENTERING THE STATEMENT.	IIIB 500
000155	NOTE B - SAVEXES 10,11, AND 12 ARE FIRST USED TO INITIALIZE	IIIB 510
000156	THE NUMBER OF BOOSTERS, ORBITERS, AND LUTS RESPECTIVELY. THEY	IIIB 520
000157	THEY ARE THEN USED AS DEFINED BELOW.	IIIB 530
000158		IIIB 540
000159		IIIB 550
000160	1 - BOOSTER AND ORBITER ARE AVAILABLE	IIIB 560
000161	2 - BOOSTER AND ORBITER ARE NOT AVAILABLE	IIIB 570
000162	3 - BOOSTER IS NOT AVAILABLE	IIIB 580
000163	4 - ORBITER IS NOT AVAILABLE	IIIB 590
000164	5 - BOOSTER AND/OR ORBITER AND/OR LUT NOT AVAILABLE	IIIB 600
000165	6 - TOTAL GENERATED LAUNCH REQUESTS	IIIB 610
000166	7 - ORBITER AND/OR LUT NOT AVAILABLE	IIIB 620

000167	* 8 - LUT NOT AVAILABLE WHEN BOOSTER AND ORBITER ARE AVAILABLE	IIIB 630
000168	* 9 -	IIIB 640
000169	* 10 - THE INPUT VARIABLE TO FUNCTION 1 (SEE NOTE B ABOVE)	IIIB 650
000170	* 11 - THE LENGTH OF THE NEXT ORBITER MISSION (SEE NOTE B ABOVE)	IIIB 660
000171	* 12 - MISSION TYPE FOR NEXT MISSION (SEE NOTE B ABOVE)	IIIB 670
000172	* 13 - TIME UNTIL NEXT LAUNCH	IIIB 680
000173	* 14 -	IIIB 690
000174	* 15 - TIME BETWEEN LAUNCH OPPORTUNITIES FOR NEXT MISSION	IIIB 700
000175	* 16 - LENGTH OF ORBITER MISSION FOR NEXT MISSION	IIIC 30
000176	* 17 - MISSION TYPE OF CURRENT MISSION	IIIB 710
000177	* 18 - TIME UNTIL CURRENT LAUNCH	IIIB 720
000178	* 19 -	IIIB 730
000179	* 20 - TIME BETWEEN LAUNCH OPPORTUNITIES FOR CURRENT MISSION	IIIB 740
000180	* 21 - TIME OF FIRST LAUNCH OPPORTUNITY FOR CURRENT MISSION	IIIB 750
000181	* 22 - VEHICLES REMOVED FROM PAD (FAILURE ON PAD)	IIIB 760
000182	* 23 - VEHICLES REMOVED FROM PAD (CONSTRAINS NEXT LAUNCH)	IIIB 770
000183	* 24 - VEHICLES REMOVED FROM PAD (NO SECOND LAUNCH WINDOW)	IIIB 780
000184	* 25 - LAUNCHES MISSED DUE TO PAD UNAVAILABILITY	IIIB 790
000185	* 26 - THE NUMBER OF VEHICLES LAUNCHED AT AN ALTERNATE LAUNCH WINDOW	IIIB 800
000186	* 27 - LAUNCH WINDOW INDICATOR	IIIC 40
000187	* 28 - NUMBER OF LAUNCHES AT PRIMARY WINDOW	IIIC 50
000188	* 29 - NUMBER OF TIMES A VEHICLE WAS REPAIRED ON THE PAD	IIIC 60
000189	* 30 - TOTAL SUCCESSFULLY ACCOMPLISHED LAUNCHES	IIIB 840
000190	* 31 -	IIIC 850
000191	* 32 -	IIIC 61
000192	* 33 -	IIIC 62
000193	* 34 -	IIIC 63
000194	* 35 - AMOUNT OF TIME AFTER PRIMARY WINDOW AT WHICH LAUNCH OCCURS	IIIC 64
000195		IIIA1500
000196		IIIA1510
000197		IIIA1520
000198		IIIA1530
000199		IIIA1540
000200		IIIA1550
000201		IIIA1560
000202	* 1 - GROUND PROCESSING TIME FOR EACH BOOSTER	IIIB 860
000203	(TIME OUT OF STORAGE)-(FLIGHT TIME)+(TIME IN STORAGE)	IIIB 870
000204		IIIB 880
000205	* 2 - TIME LEFT BEFORE LAUNCH (USED TO DETERMINE IF SUFFICIENT TIME	IIIB 890
000206	REMAINS BEFORE LAUNCH TO INITIATE AND COMPLETE COUNTDOWN)	IIIB 900
000207	(TIME TO LAUNCH FROM EARLIEST POSSIBLE STORAGE EXIT)-(ELAPSED	IIIB 910
000208	TIME FROM EARLIEST POSSIBLE STORAGE EXIT)	IIIB 920
000209	NOTE- IF FIRST LAUNCH WINDOW IS MISSED V2 WILL BE NEGATIVE.	IIIC 70
000210	THIS IS TAKEN INTO ACCOUNT IN LATER VARIABLES.	IIIC 80
000211		IIIB 930
000212	* 3 - FLOATING HOLD TIME BETWEEN COMPLETION OF ALL PRECOUNTDOWN	IIIB 940
000213	OPERATIONS AND THE INITIATION OF FINAL COUNTDOWN FOR THE	IIIB 950
000214	FIRST LAUNCH OPPORTUNITY	IIIB 960
000215	(TIME LEFT BEFORE LAUNCH)-(COUNTDOWN TIME)	IIIB 970
000216		IIIB 980
000217	* 4 - TIME TO SECOND LAUNCH WINDOW	IIIC 90
000218	(TIME FIRST WINDOW WAS MISSED BY)+(TIME BETWEEN FIRST AND	IIIC 100
000219	SECOND WINDOW)	IIIC 110
000220		IIIB1040
000221	* 5 - MAXIMUM TIME TO PROCESS VEHICLE FROM MATE AND HOOKUP INITIATION	IIIB1050
000222	TO LAUNCH. THIS ALSO REPRESENTS THE EARLIEST POSSIBLE STORAGE	IIIB1060

000223	*	EXIT FOR THE VEHICLE ELEMENTS.	IIIB1070
000224	*	(ASSIGN CONSTANT = SUM OF MAXIMUM TIMES IN EACH PRELAUNCH OPERATION)	IIIB1080
000225	*		IIIB1090
000226	*		IIIB1100
000227	*	6 - LENGTH OF COUNTDOWN	IIIB1110
000228	*	(PREASSIGNED CONSTANT)	IIIB1120
000229	*		IIIB1130
000230	*	7 - GROUND PROCESSING TIME FOR EACH ORBITER	IIIB1140
000231	*	(TIME OUT OF STORAGE)-(FLIGHT TIME)+(TIME IN STORAGE)	IIIB1150
000232	*		IIIB1160
000233	*	8 - MINIMUM TIME TO PROCESS VEHICLE FROM MATE AND HOOKUP INITIATION TO LAUNCH. THIS ALSO REPRESENTS THE LATEST POSSIBLE STORAGE EXIT FOR THE VEHICLE ELEMENTS IF THE FIRST LAUNCH WINDOW IS TO 9E MET.	IIIB1170
000234	*		IIIB1180
000235	*		IIIB1190
000236	*		IIIB1200
000237	*	ASSIGN CONSTANT = SUM OF MINIMUM TIMES IN EACH PRELAUNCH OPERATION	IIIB1210
000238	*		IIIB1220
000239	*		IIIB1230
000240	*	9 - TIME VEHICLES ARE HELD IN READY STORAGE PRIOR TO EARLIEST POSSIBLE INITIATION OF MATE AND HOOKUP RELATIVE TO THEIR MISSION (LAUNCH TIME)-(CURRENT TIME)-(VARIABLE 5)	IIIB1240
000241	*		IIIB1250
000242	*		IIIB1260
000243	*		IIIB1290
000244	*	10 - THE NUMBER OF SIMULATIONS FOR WHICH STATISTICS ARE COLLECTED (ASSIGN CONSTANT = ONE LESS THAN NUMBER OF SIMULATIONS SPECIFIED ON THE START CARD)	IIIB1300
000245	*		IIIB1310
000246	*		IIIB1320
000247	*		IIIB1330
000248	*	11 - MAXIMUM TIME A REQUEST MAY WAIT FOR ANY VEHICLE ELEMENT AFTER THE EARLIEST POSSIBLE MATE AND HOOKUP INITIATION.	IIIB1340
000249	*		IIIB1350
000250	*	(MAXIMUM TIME TO LAUNCH(V5))-(MINIMUM TIME TO LAUNCH(V8))	IIIB1360
000251	*		
000252	*	12 - TIME TO BEGINNING OF FINAL COUNTDOWN FOR THE SECOND WINDOW (TIME TO SECOND LAUNCH WINDOW)-(LENGTH OF COUNT DOWN)	IIIC 120
000253	*		IIIC 130
000254	*		IIIC 140
000255	*	13 - LENGTH OF TIME BETWEEN EVEN AND ODD WINDOWS (23.5 HOURS)-(TIME BETWEEN FIRST AND SECOND WINDOW)	IIIC 150
000256	*		IIIC 160
000257	*		IIIC 170
000258	*	14 - TIME LEFT TO BEGINNING OF FINAL COUNTDOWN FOR LAUNCH AT THE CURRENT WINDOW.	IIIC 180
000259	*		IIIC 190
000260	*	(LENGTH OF TIME BETWEEN FIRST WINDOW AND CURRENT WINDOW)+(TIME THE FIRST WINDOW WAS MISSED BY)-(LENGTH OF FINAL COUNTDOWN)	IIIC 200
000261	*		IIIC 210
000262	*		IIIC 220
000263	*	15 - DIVIDE WINDOW NUMBER BY 2 DISCARD QUOTIENT AND RETAIN REMAINDER IF V15=0 THE WINDOW IS EVEN. IF V15=1 THE WINDOW IS ODD	IIIC 230
000264	*		IIIC 240
000265	*		IIIC 250
000266	1	VARIABLE MP11-P1+P2	IIIC 260
000267	2	VARIABLE V5-M1	IIIC 270
000268	3	VARIABLE V2-V6	
000269	4	VARIABLE P7+V2	
000270	5	VARIABLE K5325	
000271	6	VARIABLE K83	
000272	7	VARIABLE MP11-P1+P4	
000273	8	VARIABLE K3925	
000274	9	VARIABLE X21-C1-V5	
000275	10	VARIABLE K500	
000276	11	VARIABLE V5-V8	
000277	12	VARIABLE V4-V6	
000278	13	VARIABLE K979-P7	

000279	14	VARIABLE	X35+V2-V6	IIIC 280
000280	15	VARIABLE	X27(K2)	IIIC 290
000281	*			IIIA1880
000282	*			IIIB1370
000283	*			IIIB1380
000284	*			IIIB1390
000285	*			IIIB1400
000286	*			IIIB1410
000287	*			IIIB1420
000288	*	1 - INDEPENDENT TOTAL FLIGHT TIME OF EACH BOOSTER AND ORBITER		IIIB1430
000289	*	2 - TIME IN STORAGE FOR THE BOOSTER OF THE CURRENT MISSION		IIIB1440
000290	*	3 - LENGTH OF CURRENT ORBITER MISSION		IIIB1450
000291	*	4 - MISSION TYPE UNTIL REDEFINED AS TIME IN STORAGE FOR THE ORBITER		IIIB1460
000292	*	OF THE CURRENT MISSION		IIIB1470
000293	*	5 - TIME TO THE CURRENT LAUNCH		IIIB1480
000294	*	6 -		IIIB1490
000295	*	7 - TIME BETWEEN LAUNCH OPPORTUNITIES FOR CURRENT LAUNCH		IIIB1500
000296	*	8 - TIME FROM LANDING TO STORAGE ENTRY		IIIB1510
000297	*	9 - DEFINES PAD TO BE SEIZED FOR CURRENT MISSION		IIIB1520
000298	*	10 - INITIALIZATION OF THE NUMBER OF BOOSTERS, ORBITERS AND LUTS		IIIB1530
000299	*	11 - IS MARKED SUCH THAT THE LEAVE STORAGE TO LAUNCH TIME CAN BE		IIIB1540
000300	*	TABULATED. THE AMOUNT OF TIME EACH BOOSTER AND ORBITER		IIIB1550
000301	*	SPENDS OUT OF STORAGE IS ALSO TABULATED BY THIS PARAMETER.		IIIB1560
000302	*			IIIB1570
000303	*			IIIA1890
000304	*			IIIA1900
000305	*			IIIA1910
000306	*	START NEW PROBLEM AND READ IN FLEET SIZE		IIIA1920
000307	*			IIIA1930
000308	1	GENERATE 0 1		IIIA1940
000309	*	BOOSTER ORBITERS	LUT	IIIA1950
000310	2	HELP K7 X11 X10	X12	IIIA1960
000311	*			IIIA1970
000312	*	ENTER X10 NUMBER OF ORBITERS INTO STORAGE 13		IIIA1980
000313	*			IIIA1990
000314	3	ASSIGN 10 X10		IIIA2000
000315	4	ENTER 13 P10		IIIA2010
000316	*			IIIA2020
000317	*	ENTER X11 NUMBER OF BOOSTERS INTO STORAGE 4		IIIA2030
000318	5	ASSIGN 10 X11		IIIA2040
000319	6	ENTER 4 P10		IIIA2050
000320	*			IIIA2060
000321	*	ENTER X12 NUMBER OF LUTS IN STORAGE 16		IIIA2070
000322	*			IIIA2080
000323	7	ASSIGN 10 X12		IIIA2090
000324	8	ENTER 16 P10		IIIB1571
000325	*			IIIA2120
000326	*			IIIA2130
000327	*			IIIA2140
000328	*			IIIA2150
000329	*			IIIA2160
000330	*	GENERATE SUFFICIENT TIME TO MEET THE FIRST MISSION REQUEST		IIIA2170
000331	*			IIIA2180
000332	*	INITIALIZE SAVEX 16 THROUGH 21 WITH FIRST TRANSACTION START TIME		IIIA2200
000333	*			IIIA2210
000334	*			IIIA2220

000335	9	SAVE	16	X11						IIIA2230
000336	10	SAVE	17	X12						IIIA2240
000337	11	SAVE	18	X13						IIIA2250
000338	12	SAVE	20	X15						IIIA2260
000339	*INITIALIZE TIME TO LAUNCH CUMULATOR WITH SIMULATION START TIME									IIIA2270
000340	13	SAVE	21	C1						IIIA2280
000341										IIIA2290
000342	* DETERMINE IF FIRST LAUNCH REQ. HAS ENOUGH TIME TO MEET ITS LAUNCH									IIIA2300
000343										IIIA2310
000344	14	COMPARE	X18	GE	V5					IIIA2320
000345										IIIA2330
000346	* CALL HELP & DETERMINE A MORE REALISTIC FIRST LAUNCH REQ.									IIIA2340
000347										IIIA2350
000348	15	HELP	K1	X11	X12	9				IIIA2360
000349										IIIA2370
000350	* RETURN AND REINITIALIZE SAVE 16 THROUGH 21									IIIA2380
000351										IIIA2390
000352	16	LOGIC	R3							IIIA2400
000353	17	TERMINATE								IIIA2410
000354										IIIA2420
000355										IIIA2430
000356										IIIA2440
000357										IIIA2450
000358										IIIA2460
000359										IIIA2470
000360										IIIA2480
000361										IIIA2490
000362										IIIA2500
000363										IIIA2510
000364	* THE SIMULATION TRANSACTIONS ARE GENERATED FOLLOWING THIS POINT									IIIA2520
000365										IIIA2530
000366										IIIA2540
000367										IIIA2550
000368										IIIA2560
000369										IIIA2570
000370										IIIA2580
000371										IIIA2590
000372										IIIA2600
000373										IIIA2610
000374										IIIA2620
000375										IIIA2630
000376	* GENERATE LAUNCHES BASED ON PROJECTED TRAFFIC REQUIREMENT									IIIA2640
000377										IIIA2650
000378										IIIA2660
000379	18	GENERATE								IIIA2670
000380	19	GATE	LR3							IIIA2680
000381	20	LOGIC	S3							IIIA2690
000382										IIIA2700
000383	*COUNT THE NUMBER OF LAUNCH REQUESTS									IIIA2710
000384										IIIA2720
000385	21	SAVE	S+	K1						IIIA2730
000386										IIIA2740
000387	*CALL HELP TO GENERATE ATTRIBUTES OF NEXT LAUNCH									IIIA2750
000388										IIIA2760
000389	22	HELP	K1	X11	X12	23				IIIA2770
000390										IIIA2780

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000447	51	COMPARE	021	E	K0			52		IIIA3310	
000448	52	LEAVE	31					53		IIIC 300	
000449	53	TABULATE	6					54		IIIC 310	
000450	*									IIIA3330	
000451	*									IIIA3340	
000452	*									IIIA3350	
000453	*									IIIA3360	
000454	*	COUNT THE NUMBER OF TIMES THE BOOSTER, ORBITER AND LUT ARE AVAILABLE								IIIA3370	
000455	*	REMOVE ONE BOOSTER, ORBITER, AND LUT FROM THEIR RESPECTIVE STORAGES,								IIIB1620	
000456	*	AND TABULATE THE TIME THESE VEHICLE ELEMENTS SPENT IN STORAGE.								IIIB1630	
000457	*									IIIA3380	
000458	54	SAVE	1+	K1				55		IIIA3390	
000459	55	LEAVE	4					56		IIIA3400	
000460	56	HELP	K6	K1	C1			57	X10	IIIA3410	
000461	57	TABULATE	2					58		IIIA3420	
000462	58	ASSIGN	2	X10				59		IIIA3430	
000463	59	LEAVE	13					60		IIIA3440	
000464	60	HELP	K6	K2	C1			61	X10	IIIA3450	
000465	61	TABULATE	22					62		IIIA3460	
000466	62	ASSIGN	4	X10				63		IIIA3470	
000467	63	LEAVE	16					64		IIIA3480	
000468	*									IIIA3490	
000469	*	PARAMETER 11 MEASURES LEAVE STORAGE TO LAUNCH TIME								IIIA3500	
000470	64	MARK	11					65		IIIA3510	
000471	*									IIIA3520	
000472	*									IIIA3530	
000473	*									IIIA3540	
000474	*	ENTER-WAITING LINE FOR MATE AND HOOKUP FACILITY								IIIA3550	
000475	*									IIIA3560	
000476	65	QUEUE	20					66		IIIA3570	
000477	*MATE	BOOSTER, ORBITER AND LUT								IIIA3580	
000478	66	ENTER	20					67		IIIA3590	
000479	*	TRIANGULAR		MIN	MODE				MAX	IIIA3600	
000480	67	HELP	K3	K1800	K2000			68	K2200 X10	IIIC 311	
000481	68	ADVANCE						69	1 FN1	IIIA3620	
000482	69	LEAVE	20					70		IIIA3630	
000483	*									IIIA3640	
000484	*									IIIA3650	
000485	*									IIIA3660	
000486	*									IIIA3680	
000487	*	WAITING LINE FOR PAD								IIIA3690	
000488	*									IIIA3700	
000489	70	QUEUE	21					71	72	IIIA3710	
000490	71	COMPARE	F1	E	K0			74		IIIA3720	
000491	72	COMPARE	F2	E	K0			75		IIIA3730	
000492	73	COMPARE	F3	E	K0			76		IIIA3740	
000493	74	ASSIGN	9	K1				77		IIIA3750	
000494	75	ASSIGN	9	K2				77		IIIA3760	
000495	76	ASSIGN	9	K3				77		IIIA3770	
000496	*									IIIA3780	
000497	*									IIIA3790	
000498	*									IIIA3800	
000499	*									IIIA3810	
000500	*	SEIZE THE FIRST AVAILABLE PAD FOR THIS MISSION AND								IIIA3820	
000501	*	REMOVE THIS PAD FROM THE PAD AVAILABLE LIST								IIIA3830	
000502	*									IIIA3840	

000503	77	SEIZE	*9	78		IIIA3850
000504	*					IIIA3860
000505	*					IIIA3880
000506	*					IIIA3890
000507	*					IIIA3900
000508	*	COUNTDOWN PREPARATION				IIIA3930
000509	*	TRANSFER LUT AND VEHICLE, INSTALL INFO PAD AND COUNTDOWN PREPS				IIIA3920
000510	*					IIIA3940
000511	78	ENTER	22	79		IIIA3950
000512	*	TRIANGULAR	MIN	MODE	MAX	IIIC 312
000513	79	HELP	K3	K2000	K2500	IIIA3970
000514	80	ADVANCE		81	FN1	IIIC 320
000515	81	LEAVE	22	82		IIIA3990
000516	*					IIIA4000
000517	*					IIIA4010
000518	*					IIIC 330
000519	*					IIIC 340
000520	*	DETERMINE WHICH PATH THROUGH THE LAUNCH LOGIC IS TO BE TAKEN				IIIC 350
000521	*	ESTABLISH TIME LEFT TO FIRST LAUNCH WINDOW				IIIC 360
000522	82	SAVEX	10	V2	ALL	IIIC 370
000523	*					IIIC 380
000524	*					IIIC 390
000525	*					IIIC 400
000526	*	IS TIME TO LAUNCH GE LENGTH OF FINAL COUNTDOWN. IF SO GO TO				IIIC 410
000527	*	PRIMARY PATH				IIIC 420
000528	*					IIIC 430
000529	*PATH ONE					IIIC 440
000530	83	COMPARE	V2	GE	V6	IIIC 450
000531	*					IIIC 460
000532	*	TIME TO LAUNCH IS NOT GE LENGTH OF FINAL COUNTDOWN, IS THERE				IIIC 470
000533	*	ANOTHER LAUNCH WINDOW. IF SO GO TO SECONDARY PATH				IIIC 480
000534	*					IIIC 490
000535	*PATH TWO					IIIC 500
000536	84	COMPARE	P7	G	K0	IIIC 510
000537	*					IIIC 520
000538	*	TIME TO LAUNCH IS NOT GE LENGTH OF FINAL COUNTDOWN, AND THERE				IIIC 530
000539	*	IS NOT ANOTHER LAUNCH WINDOW. COUNT NUMBER OF VEHICLES REMOVED				IIIC 540
000540	*	FROM THE PAD FOR LACK OF ANOTHER LAUNCH WINDOW AND GO TO THE				IIIC 550
000541	*	VEHICLE REMOVAL PATH.				IIIC 560
000542	*					IIIC 570
000543	*PATH THREE					IIIC 580
000544	85	SAVEX	24	K1	210	IIIC 590
000545	*					IIIC 600
000546	*					IIIC 610
000547	*					IIIC 620
000548	*					IIIC 630
000549	*	THIS IS THE PRIMARY PATH THROUGH THE LAUNCH LOGIC.				IIIC 640
000550	*					IIIC 650
000551	*					IIIC 660
000552	*					IIIC 670
000553	*					IIIC 680
000554	*	ESTABLISH LENGTH OF HOLD TO BEGINNING OF FINAL COUNTDOWN				IIIC 690
000555	86	SAVEX	10	V3	87	IIIC 700
000556	*	TABULATE LENGTH OF HOLD TO BEGINNING OF FINAL COUNTDOWN				IIIC 710
000557	87	TABULATE	4	88		IIIC 720
000558	*	SET FIRST LAUNCH WINDOW INDICATOR				

000559	88	SAVEX	27	K1	89	IIIC 730
000560	*	ESTABLISH A BASE POINT OF ZERO FOR CALCULATING TIME BETWEEN				IIIC 740
000561	*	FIRST AND SUBSEQUENT WINDOWS IN CASE THIS VEHICLE HAS A FINAL				IIIC 750
000562	*	COUNTDOWN FAILURE AND CAN BE REPAIRED ON THE PAD.				IIIC 760
000563	89	SAVEX	35	K1	90	IIIC 770
000564	*	ADVANCE TO BEGINNING OF FINAL COUNTDOWN				IIIC 780
000565	90	ADVANCE		1	FN1	IIIC 790
000566	*					IIIC 800
000567	*					IIIC 810
000568	*					IIIC 820
000569	*	FINAL COUNTDOWN				IIIC 830
000570	*					IIIC 840
000571	*					IIIC 850
000572	*	BEGIN FINAL COUNTDOWN				IIIC 860
000573	*					IIIC 870
000574	91	ENTER	23			IIIC 880
000575	*	ESTABLISH LENGTH OF FINAL COUNTDOWN				IIIC 890
000576	92	HELP	K11	K83	X10	IIIC 900
000577	*	ADVANCE TO END OF FINAL COUNTDOWN				IIIC 910
000578	93	ADVANCE		1	FN1	IIIC 920
000579	*	LEAVE FINAL COUNTDOWN				IIIC 930
000580	94	LEAVE	23	.05	100	IIIC 940
000581	*					IIIC 950
000582	*					IIIC 960
000583	*					IIIC 970
000584	*					IIIC 980
000585	*	FAILURE HAS NOT OCCURRED DURING FINAL COUNTDOWN. COLLECT				IIIC 990
000586	*	APPROPRIATE STATISTICS AND INITIATE LAUNCH OF VEHICLE.				IIIC 995
000587	*	(95 PERCENT PROBABILITY)				IIIC1000
000588	*					IIIC1010
000589	*	TABULATE WHICH WINDOW LAUNCH OCCURS AT				IIIC1020
000590	95	TABULATE	9	BOTH	96	IIIC1030
000591	*	DID LAUNCH OCCUR AT PRIMARY WINDOW				IIIC1040
000592	96	COMPARE	X27	E	K1	IIIC1050
000593	*	COUNT NUMBER OF LAUNCHES AT THE PRIMARY WINDOW				IIIC1060
000594	97	SAVEX	28+	K1	118	IIIC1070
000595	*	LAUNCH DID NOT OCCUR AT THE PRIMARY WINDOW				IIIC1080
000596	*	TABULATE AMOUNT OF TIME AFTER PRIMARY WINDOW LAUNCH OCCURRED				IIIC1090
000597	98	TABULATE	8		99	IIIC1100
000598	*	COUNT NUMBER OF LAUNCHES AT A SUBSEQUENT WINDOW				IIIC1110
000599	99	SAVEX	26+	K1	118	IIIC1120
000600	*					IIIC1130
000601	*					IIIC1140
000602	*					IIIC1150
000603	*	FAILURE HAS OCCURRED DURING FINAL COUNTDOWN (5 PERCENT PROB.)				IIIC1160
000604	*	REMOVE 50 PERCENT OF THESE VEHICLES FROM PAD. ATTEMPT TO LAUNCH				IIIC1170
000605	*	50 PERCENT AT A LATER WINDOW.				IIIC1180
000606	100	ADVANCE		.50	101	IIIC1190
000607	*	COUNT NUMBER OF VEHICLES REMOVED FROM PAD DUE TO A FINAL			102	IIIC1200
000608	*	COUNTDOWN FAILURE.				IIIC1210
000609	101	SAVEX	22+	K1	210	IIIC1220
000610	*	COUNT NUMBER OF TIMES A VEHICLE WAS REPAIRED ON THE PAD				IIIC1230
000611	102	SAVEX	29+	K1	BOTH 103	IIIC1240
000612	*	IS THERE ANOTHER LAUNCH OPPORTUNITY			104	IIIC1250
000613	103	COMPARE	P7	G	K0	IIIC1260
000614	*				112	IIIC1270

000615	*	THERE IS NOT ANOTHER LAUNCH OPPORTUNITY	IIIC1280
000616	*	COUNT NUMBER OF VEHICLES REMOVED DUE TO NO FURTHER	IIIC1290
000617	*	LAUNCH WINDOW. THEN GO TO VEHICLE REMOVAL PATH.	IIIC1300
000618	104	SAVEX 24+ K1	IIIC1310
000619	*	THERE IS ANOTHER LAUNCH OPPORTUNITY	IIIC1320
000620	*	SET SECOND LAUNCH WINDOW INDICATOR	IIIC1330
000621	*		IIIC1340
000622	*		IIIC1350
000623	*		IIIC1360
000624	*		IIIC1370
000625	*		IIIC1380
000626	*	THIS THE SECONDARY PATH THROUGH THE LAUNCH LOGIC. ONLY VEHICLES	IIIC1390
000627	*	WHICH CANNOT LAUNCH ON THE PRIMARY WINDOW FLOW THROUGH THIS	IIIC1400
000628	*	SECTION	IIIC1405
000629	*		IIIC1410
000630	105	SAVEX 27 K2	IIIC1420
000631	*	TABULATE AMOUNT OF TIME LEFT TO FIRST WINDOW (MAY BE NEGATIVE).	IIIC1430
000632	106	TABULATE 7 BOTH 107 111	IIIC1440
000633	*	IS THERE TIME TO LAUNCH AT THE SECOND WINDOW	IIIC1450
000634	107	COMPARE V4 GE V6	IIIC1460
000635	*	THERE IS TIME TO LAUNCH AT THE SECOND WINDOW	IIIC1470
000636	*	ESTABLISH LENGTH OF HOLD TO BEGINNING OF FINAL COUNTDOWN	IIIC1480
000637	*	FOR THE SECOND WINDOW.	IIIC1490
000638	108	SAVEX 10 V12	IIIC1500
000639	*	ESTABLISH LENGTH OF TIME BETWEEN FIRST AND SECOND WINDOW	IIIC1510
000640	109	SAVEX 35 P7	IIIC1520
000641	*	TABULATE HOLD TO BEGINNING OF FINAL COUNTDOWN FOR THE	IIIC1530
000642	*	SECOND WINDOW	IIIC1540
000643	110	TABULATE 5	IIIC1550
000644	*		IIIC1560
000645	*		IIIC1570
000646	*	THERE IS NOT TIME TO LAUNCH AT THE SECOND WINDOW	IIIC1580
000647	*	ESTABLISH LENGTH OF TIME BETWEEN FIRST AND SECOND WINDOW	IIIC1590
000648	111	SAVEX 35 P7	IIIC1600
000649	*	INCREMENT WINDOW INDICATOR TO NEXT WINDOW	IIIC1610
000650	112	SAVEX 27+ K1 BOTH 113 115	IIIC1620
000651	*	DETERMINE IF THE WINDOW BEING TRIED IS EVEN OR ODD	IIIC1630
000652	113	COMPARE V15 G K0	IIIC1640
000653	*	THE WINDOW IS ODD.	IIIC1650
000654	*	ESTABLISH LENGTH OF TIME BETWEEN FIRST AND SUBSEQUENT	IIIC1660
000655	114	SAVEX 35+ V13	IIIC1670
000656	*	ODD WINDOWS	IIIC1680
000657	*		IIIC1690
000658	*	THE WINDOW IS EVEN	IIIC1700
000659	*	ESTABLISH LENGTH OF TIME BETWEEN FIRST AND SUBSEQUENT	IIIC1710
000660	115	SAVEX 35+ P7	IIIC1720
000661	*	EVEN WINDOWS	IIIC1730
000662	116	SAVEX 35+ P7	IIIC1740
000663	*	IS THERE TIME TO LAUNCH AT THIS WINDOW. IF NOT TRY NEXT WINDOW.	IIIC1750
000664	117	COMPARE V14 GE K0	IIIC1760
000665	*	THERE IS TIME TO LAUNCH	IIIC1770
000666	*	ESTABLISH LENGTH OF HOLD TO BEGINNING OF FINAL COUNTDOWN	IIIC1780
000667	118	SAVEX 10 V14	IIIC1790
000668	*		IIIC1800
000669	*		IIIC1810
000670	*		IIIC1820

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000727	*	FLIGHT AND REFURBISH PATH							IIIC1910
000728	*								IIIA5470
000729	*								IIIA5480
000730	*	142 TABULATE	10						IIIA5490
000731	*								IIIA5500
000732	*								IIIA5510
000733	*	*LIFTOFF THROUGH SEPARATION							IDST
000734	*								IIIA5520
000735	*	143 ENTER	28						IIIA5530
000736	*	144 HELP	K11 K2						IIIA5540
000737	*	145 ADVANCE							IIIA5550
000738	*	146 LEAVE	28						IIIA5560
000739	*								IIIA5570
000740	*								IIIA5580
000741	*								IIIA5590
000742	*								IIIA5600
000743	*								IIIA5610
000744	*	*PARAMETER 1 IS USED TO CUMULATE THE TOTAL INFLIGHT TIME FOR							IIIA5620
000745	*	*COLLECTING STATISTICS ON GROUND PROCESSING TIME							IIIA5630
000746	*								IIIA5640
000747	*	147 ASSIGN	1+ K2						IIIA5650
000748	*								IIIA5660
000749	*	*SEPARATION IN FLIGHT OF BOOSTER AND ORBITER							IIIA5670
000750	*	*COUNT THE NUMBER OF SUCCESSFUL LAUNCHES							IIIA5680
000751	*								IIIA5690
000752	*	148 SAVEX	30+ K1						IIIA5700
000753	*	149 SPLIT							IIIA5710
000754	*								IIIA5720
000755	*								IIIA5730
000756	*								IIIA5740
000757	*								IIIA5750
000758	*								IIIA5760
000759	*	*BOOSTER COAST, REENTRY, FLYBACK AND LANDING							IIIA5770
000760	*								IIIA5780
000761	*	150 HELP	K3 K64 K70						IIIA5790
000762	*	151 ASSIGN	1+ FN1						IIIA5800
000763	*	152 ADVANCE							IIIA5810
000764	*								IIIA5820
000765	*								IIIA5830
000766	*								IIIA5840
000767	*	*PARAMETER 8 COLLECTS THE TIME FROM LANDING UNTIL ENTRY INTO STORAGE							IIIA5850
000768	*	153 MARK	8						IIIA5860
000769	*								IIIA5870
000770	*								IIIA5880
000771	*								IIIA5890
000772	*								IIIA5900
000773	*								IIIA5910
000774	*								IIIA5920
000775	*	*DELAY FOR TRANSPORTING FROM DOWNRANGE LANDING							IIIA5930
000776	*								IIIA5940
000777	*								IIIA5950
000778	*	154 HELP	K3 K1000 K1500						IIIA5960
000779	*	155 SAVEX	10 K0						IIIA5970
000780	*	156 ADVANCE							IIIA5980
000781	*								IIIA5990
000782	*								IIIA6000
									IIIA6010

000783	*						IIIA6020
000784	*						IIIA6030
000785	*	* WAITING LINE FOR BOOSTERS GOING INTO SAFING FACILITY					IIIA6040
000786	*						IIIA6050
000787	*	157	QUEUE	1			IIIA6060
000788	*						IIIA6070
000789	*						IIIA6080
000790	*						IIIA6090
000791	*						IIIA6100
000792	*	* BOOSTER SAFING					IDST
000793	*						IIIA6110
000794	*	158	ENTER	1			IIIA6120
000795	*		TRIANGULAR	K3	MIN	MODE	IIIA6130
000796	*	159	HELP				IIIC1911
000797	*	160	ADVANCE				IIIA6150
000798	*	161	LEAVE	1			IIIA6160
000799	*						IIIA6170
000800	*						IIIA6180
000801	*						IIIA6190
000802	*	* TRANSPORT BOOSTER TO MAINTENANCE					IIIA6200
000803	*						IDST
000804	*		TRIANGULAR	K3	MIN	MODE	IIIA6220
000805	*	162	HELP				IIIC1912
000806	*	163	ADVANCE				IIIA6240
000807	*						IIIA6250
000808	*						IIIA6260
000809	*						IIIA6270
000810	*						IIIA6280
000811	*	* BOOSTER MAINTENANCE					IIIA6290
000812	*						IDST
000813	*	164	QUEUE	2			IIIA6310
000814	*	165	ENTER	2			IIIA6320
000815	*		TRIANGULAR	K3	MIN	MODE	IIIA6330
000816	*	166	HELP				IIIC1917
000817	*	167	ADVANCE				IIIA6350
000818	*						IIIA6360
000819	*						IIIA6370
000820	*	* BOOSTER ADDITIONAL MAINTENANCE					IIIA6380
000821	*		EXPONENTIAL	K8	MIN	MEAN	IIIA6390
000822	*	168	HELP				IIIC1913
000823	*	169	ADVANCE				IIIA6410
000824	*	170	LEAVE	2			IIIA6420
000825	*						IIIA6430
000826	*						IIIA6440
000827	*						IIIA6450
000828	*						IIIA6460
000829	*	* TRANSPORT BOOSTER TO STORAGE					IIIA6470
000830	*						IDST
000831	*	171	QUEUE	3			IIIA6490
000832	*	172	ENTER	3			IIIA6500
000833	*	173	HELP	K3	K1900	K2000	IIIA6510
000834	*	174	ADVANCE				IIIA6520
000835	*	175	LEAVE	3			IIIA6530
000836	*						IIIA6540
000837	*	* BOOSTER STORAGE WHEN WAITING FOR ORBITER OR LAUNCH REQUIREMENT					IIIA6550
000838	*						IIIA6560
							IIIA6570

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000839	176 ENTER	4	K1		177	IIIA6580
000840	177 HELP	5	K1	C1	178	IIIA6590
000841	178 TABULATE	3			179	IIIA6600
000842	179 TABULATE	1			260	IIIA6610
000843	*					IIIA6620
000844	*					IIIA6630
000845	*					IIIA6640
000846	* ORBITER BOOST INTO ORBIT,PERFORM MISSION AND RETURN					IIIA6650
000847	*					IIIA6660
000848	180 ENTER	14			181	IIIA6670
000849	181 SAVE	10	P3		182	IIIA6680
000850	182 ASSIGN	1+	FN1		183	IIIA6690
000851	183 ADVANCE				184	IIIA6700
000852	184 LEAVE	14			185	IIIA6710
000853	*					IIIA6720
000854	*					IIIA6730
000855	*					IIIA6740
000856	*PARAMETER 8 COLLECTS THE TIME FROM LANDING UNTIL ENTRY INTO STORAGE					IIIA6750
000857	185 MARK	8			186	IIIA6760
000858	*					IIIA6770
000859	*					IIIA6780
000860	*					IIIA6790
000861	*					IIIA6800
000862	*ORBITER SAFING					IDST IIIA6810
000863	*					IIIA6820
000864	186 QUEUE	10			187	IIIA6830
000865	187 ENTER	10			188	IIIA6840
000866	* TRIANGULAR		MIN	MODE		IIIC1914
000867	188 HELP	K3	K900	K1000	189	IIIA6860
000868	189 ADVANCE				190	IIIA6870
000869	190 LEAVE	10			191	IIIA6880
000870	*					IIIA6890
000871	*					IIIA6900
000872	*					IIIA6910
000873	*					IIIA6920
000874	*TRANSPORT ORBITER TO MAINTENANCE					IDST IIIA6930
000875	*					IIIA6940
000876	* LOGNORMAL		MEAN	STDDEV		IIIA6950
000877	191 HELP	K3	K50	K100	192	IIIA6960
000878	192 ADVANCE				193	IIIA6970
000879	*					IIIA6980
000880	*					IIIA6990
000881	*					IIIA7000
000882	*					IIIA7010
000883	*ORBITER MAINTENANCE					IDST IIIA7020
000884	*					IIIA7030
000885	193 QUEUE	11			194	IIIA7040
000886	194 ENTER	11			195	IIIA7050
000887	* TRIANGULAR		MIN	MODE		IIIC1915
000888	195 HELP	K3	K2250	K2500	196	IIIA7070
000889	196 ADVANCE				197	IIIA7080
000890	*				25	IIIA7090
000891	*ORBITER ADDITIONAL MAINTENANCE					IDST IIIA7100
000892	*					IIIA7110
000893	* EXPONENTIAL		MIN	MEAN		IIIC1916
000894	197 HELP	K8	K500	K1000	198	IIIA7130

000895	198	ADVANCE		199	1	FM1	IIIA7140
000896	199	LEAVE	11	200			IIIA7150
000897	*	-----					IIIA7160
000898	*						IIIA7170
000899	*						IIIA7180
000900	*						IIIA7190
000901	*	TRANSPORT ORBITER TO SYSTEMS TEST, TEST, TRANSPORT TO STORAGE					IIIA7200
000902	*	ORBITER VEHICLE SYSTEM TEST FACILITY					IIIA7210
000903	*						IIIA7220
000904	200	QUEUE	12	201			IIIA7230
000905	201	ENTER	12	202			IIIA7240
000906	202	HELP	K3	K1900 K2000	K2100 X10		IIIA7250
000907	203	ADVANCE		204	1	FM1	IIIA7260
000908	204	LEAVE	12	205			IIIA7270
000909	*						IIIA7280
000910	*	ORBITER STORAGE WHEN WAITING FOR BOOSTER OR LAUNCH REQUIREMENT					IIIA7290
000911	*						IIIA7300
000912	205	ENTER	13	K1	206		IIIA7310
000913	206	HELP	K5	K2	C1		IIIA7320
000914	207	TABULATE	21		208		IIIA7330
000915	208	TABULATE	23		209		IIIA7340
000916	209	TERMINATE					IIIA7350
000917	*	-----					IIIA7360
000918	*						IIIA7370
000919	*						IIIA7380
000920	*	-----					IIIC1920
000921	*	VEHICLE REMOVAL PATH					IIIC1930
000922	*	ONLY VEHICLES WHICH ARE TO BE REMOVED FROM THE PAD FLOW					IIIC1940
000923	*	THROUGH THIS SECTION.					IIIC1950
000924	*						IIIC1960
000925	*						IIIC1970
000926	*	REMOVE BOOSTER, ORBITER AND LUT FROM THE PAD					IIIC1980
000927	*						IIIC1990
000928	210	ENTER	25	211			IIIC2000
000929	*	TRIANGULAR			MAX		IIIC2010
000930	211	HELP	K3	K1800 K2000	K2200 X10		IIIC2020
000931	212	ADVANCE		213	1	FM1	IIIC2030
000932	213	LEAVE	25	214			IIIC2040
000933	214	RELEASE	*9	215			IIIC2050
000934	*	-----					IIIC2060
000935	*						IIIC2070
000936	*						IIIC2080
000937	*	-----					IIIC2090
000938	*	TRANSPORT VEHICLE, SEPARATE ORBITER, TRANSPORT ORBITER TO MAINT					IIIC2100
000939	*						IIIC2110
000940	215	ENTER	30	216			IIIC2120
000941	*	TRIANGULAR			MAX		IIIC2130
000942	216	HELP	K3	K550 K750	K950 X10		IIIC2140
000943	217	ADVANCE		218	1	FM1	IIIC2150
000944	218	LEAVE	30	219			IIIC2160
000945	*	-----					IIIC2170
000946	*						IIIC2180
000947	*						IIIC2190
000948	*	-----					IIIC2200
000949	*						IIIC2210
000950	*	SEPARATE ORBITER FROM BOOSTER/LUT AND SEND ORBITER TO MAINTENANCE					IIIC2220

001007	247 ENTER	13		248	IIIA7820
001008	248 HELP	K5	K2 C1	249	IIIA7830
001009	249 TERMINATE				IIIA7840
001010	*				IIIC2370
001011	*				IIIA7860
001012	*				IIIA7870
001013	*				IIIA7880
001014	*				IIIA7890
001015	*				IIIA7900
001016	*				IIIA7910
001017	250 SAVEX	5+ K1	BOTH 251 255		IIIA7920
001018	251 GATE	SE4	BOTH 252 254		IIIA7930
001019	252 GATE	SE13	253		IIIA7940
001020	*				IIIA7950
001021	*NO BOOSTER OR ORBITER AVAILABLE				IIIA7960
001022	*				IIIA7970
001023	253 SAVEX	2+ K1	250		IIIA7980
001024	*				IIIA7990
001025	*NO BOOSTER AVAILABLE				IIIA8000
001026	*				IIIA8010
001027	254 SAVEX	3+ K1	260		IIIA8020
001028	*				IIIA8030
001029	*				IIIA8040
001030	*				IIIA8050
001031	*				IIIA8060
001032	255 SAVEX	7+ K1	BOTH 256 258		IIIA8070
001033	256 GATE	SE13	257		IIIA8080
001034	*				IIIA8090
001035	* NO ORBITER AVAILABLE				IIIA8100
001036	*				IIIA8110
001037	257 SAVEX	4+ K1	260		IIIA8120
001038	*				IIIA8130
001039	*				IIIA8140
001040	*				IIIA8150
001041	258 SAVEX	8+ K1	260		IIIA8160
001042	*				IIIA8170
001043	* COUNT NUMBER OF MISSED LAUNCHES PAQ(S) UNAVAILABLE				IIIA8180
001044	259 SAVEX	25+ K1	260		IIIA8190
001045	*				IIID 30
001046	*				IIID 40
001047	*				IIID 50
001048	* INSERT HELP CARDS IN THIS BLOCK TO REQUEST CONFIDENCE INTERVALS.				IIID 60
001049	* THESE ARE THE FIVE DUMMY ADVANCE CARDS USED AS PLACE HOLDERS AND				IIID 70
001050	* ARE REPLACED BY HELP CARDS WHEN REQUESTING CONFIDENCE INTERVALS.				IIID 80
001051	*				IIID 90
001052	260 ADVANCE		261		IIID 100
001053	261 ADVANCE		262		IIID 110
001054	262 ADVANCE		263		IIID 120
001055	263 ADVANCE		264		IIID 130
001056	264 ADVANCE		265		IIID 140
001057	*				IIID 150
001058	*				IIID 160
001059	*				IIID 170
001060	265 COMPARE	N276 LE	V10 BOTH 266 276		IIIA8200
001061	266 COMPARE	V10 E	N276 267		IIIA8210
001062	*				IIIA8220

* PRINT OUT SUMMARY TABLE

001063	*	001063	267	HELP	K10	268	269	270	271	272	273	274	275	276	277	278	279	280	281	282	283	284	285	286	287	288	289	290	291	292	293	294	295	296	297	298	299	300	301	302	303	304	305	306	307	308	309	310	311	312	313	314	315	316	317	318	319	320	321	322	323	324	325	326	327	328	329	330	331	332	333	334	335	336	337	338	339	340	341	342	343	344	345	346	347	348	349	350	351	352	353	354	355	356	357	358	359	360	361	362	363	364	365	366	367	368	369	370	371	372	373	374	375	376	377	378	379	380	381	382	383	384	385	386	387	388	389	390	391	392	393	394	395	396	397	398	399	400	401	402	403	404	405	406	407	408	409	410	411	412	413	414	415	416	417	418	419	420	421	422	423	424	425	426	427	428	429	430	431	432	433	434	435	436	437	438	439	440	441	442	443	444	445	446	447	448	449	450	451	452	453	454	455	456	457	458	459	460	461	462	463	464	465	466	467	468	469	470	471	472	473	474	475	476	477	478	479	480	481	482	483	484	485	486	487	488	489	490	491	492	493	494	495	496	497	498	499	500	501	502	503	504	505	506	507	508	509	510	511	512	513	514	515	516	517	518	519	520	521	522	523	524	525	526	527	528	529	530	531	532	533	534	535	536	537	538	539	540	541	542	543	544	545	546	547	548	549	550	551	552	553	554	555	556	557	558	559	560	561	562	563	564	565	566	567	568	569	570	571	572	573	574	575	576	577	578	579	580	581	582	583	584	585	586	587	588	589	590	591	592	593	594	595	596	597	598	599	600	601	602	603	604	605	606	607	608	609	610	611	612	613	614	615	616	617	618	619	620	621	622	623	624	625	626	627	628	629	630	631	632	633	634	635	636	637	638	639	640	641	642	643	644	645	646	647	648	649	650	651	652	653	654	655	656	657	658	659	660	661	662	663	664	665	666	667	668	669	670	671	672	673	674	675	676	677	678	679	680	681	682	683	684	685	686	687	688	689	690	691	692	693	694	695	696	697	698	699	700	701	702	703	704	705	706	707	708	709	710	711	712	713	714	715	716	717	718	719	720	721	722	723	724	725	726	727	728	729	730	731	732	733	734	735	736	737	738	739	740	741	742	743	744	745	746	747	748	749	750	751	752	753	754	755	756	757	758	759	760	761	762	763	764	765	766	767	768	769	770	771	772	773	774	775	776	777	778	779	780	781	782	783	784	785	786	787	788	789	790	791	792	793	794	795	796	797	798	799	800	801	802	803	804	805	806	807	808	809	810	811	812	813	814	815	816	817	818	819	820	821	822	823	824	825	826	827	828	829	830	831	832	833	834	835	836	837	838	839	840	841	842	843	844	845	846	847	848	849	850	851	852	853	854	855	856	857	858	859	860	861	862	863	864	865	866	867	868	869	870	871	872	873	874	875	876	877	878	879	880	881	882	883	884	885	886	887	888	889	890	891	892	893	894	895	896	897	898	899	900	901	902	903	904	905	906	907	908	909	910	911	912	913	914	915	916	917	918	919	920	921	922	923	924	925	926	927	928	929	930	931	932	933	934	935	936	937	938	939	940	941	942	943	944	945	946	947	948	949	950	951	952	953	954	955	956	957	958	959	960	961	962	963	964	965	966	967	968	969	970	971	972	973	974	975	976	977	978	979	980	981	982	983	984	985	986	987	988	989	990	991	992	993	994	995	996	997	998	999	1000	1001	1002	1003	1004	1005	1006	1007	1008	1009	1010	1011	1012	1013	1014	1015	1016	1017	1018	1019	1020	1021	1022	1023	1024	1025	1026	1027	1028	1029	1030	1031	1032	1033	1034	1035	1036	1037	1038	1039	1040	1041	1042	1043	1044	1045	1046	1047	1048	1049	1050	1051	1052	1053	1054	1055	1056	1057	1058	1059	1060	1061	1062	1063	1064	1065	1066	1067	1068	1069	1070	1071	1072	1073	1074	1075	1076	1077	1078	1079	1080	1081	1082	1083	1084	1085	1086	1087	1088	1089	1090	1091	1092	1093	1094	1095	1096	1097	1098	1099	1100	1101	1102	1103	1104	1105	1106	1107	1108	1109	1110	1111	1112	1113	1114	1115	1116	1117	1118	1119	1120	1121	1122	1123	1124	1125	1126	1127	1128	1129	1130	1131	1132	1133	1134	1135	1136	1137	1138	1139	1140	1141	1142	1143	1144	1145	1146	1147	1148	1149	1150	1151	1152	1153	1154	1155	1156	1157	1158	1159	1160	1161	1162	1163	1164	1165	1166	1167	1168	1169	1170	1171	1172	1173	1174	1175	1176	1177	1178	1179	1180	1181	1182	1183	1184	1185	1186	1187	1188	1189	1190	1191	1192	1193	1194	1195	1196	1197	1198	1199	1200	1201	1202	1203	1204	1205	1206	1207	1208	1209	1210	1211	1212	1213	1214	1215	1216	1217	1218	1219	1220	1221	1222	1223	1224	1225	1226	1227	1228	1229	1230	1231	1232	1233	1234	1235	1236	1237	1238	1239	1240	1241	1242	1243	1244	1245	1246	1247	1248	1249	1250	1251	1252	1253	1254	1255	1256	1257	1258	1259	1260	1261	1262	1263	1264	1265	1266	1267	1268	1269	1270	1271	1272	1273	1274	1275	1276	1277	1278	1279	1280	1281	1282	1283	1284	1285	1286	1287	1288	1289	1290	1291	1292	1293	1294	1295	1296	1297	1298	1299	1300	1301	1302	1303	1304	1305	1306	1307	1308	1309	1310	1311	1312	1313	1314	1315	1316	1317	1318	1319	1320	1321	1322	1323	1324	1325	1326	1327	1328	1329	1330	1331	1332	1333	1334	1335	1336	1337	1338	1339	1340	1341	1342	1343	1344	1345	1346	1347	1348	1349	1350	1351	1352	1353	1354	1355	1356	1357	1358	1359	1360	1361	1362	1363	1364	1365	1366	1367	1368	1369	1370	1371	1372	1373	1374	1375	1376	1377	1378	1379	1380	1381	1382	1383	1384	1385	1386	1387	1388	1389	1390	1391	1392	1393	1394	1395	1396	1397	1398	1399	1400	1401	1402	1403	1404	1405	1406	1407	140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<<<SOS MODEL UPDATE MSFC VERSION III-D>>>

001119	9	TABLE	X27	000	1	15	IIIC2540
001120	10	TABLE	MP11	0000	500	30	IIIA8380
001121	*11						IIIC2550
001122	*12						IIIC2560
001123	*13						IIIC2570
001124	*14						IIIC2580
001125	*15						IIIC2590
001126	*16						IIIC2600
001127	*17						IIIC2610
001128	*18						IIIC2620
001129	*19						IIIC2630
001130	*20						IIIC2640
001131	21	TABLE	MP8	5000	500	20	IIIA8390
001132	22	TABLE	X10	000	500	30	IIIA8400
001133	23	TABLE	V7	9000	500	30	IIIA8410
001134	*						IIIB1940
001135	*						IIIB1950
001136	*						IIIB1960
001137	*						IIIB1970
001138	*						IIIB1980
001139	*						IIIB1990
001140	NEW						
		START		501			

NUMBER OF SIMULATIONS TO BE PERFORMED

NOTE- SEE VARIABLE 10

APPENDIX B. SUPPLEMENTAL FORTRAN INSTRUCTIONS DESCRIPTION AND LISTING

As the SOS Model evolved, it became apparent that additional Fortran subprograms were required to supplement the basic instructions provided by the standard GPSS II language. These subprograms, which are referred to as the user written routines, are utilized during the dynamic simulation of the Shuttle operations cycle by means of the GPSS II HELP block. The subprograms utilized in the SOS Model are listed after the descriptions in this Appendix. The function of the Fortran subprogram HELP is described below:

- HELP serves as the interfacing routine between the GPSS II instructions and the user written subprograms. The HELP subprogram is accessed by a GPSS II HELP block containing a control variable located in the X field. The HELP subprogram serves four general functions in support of the SOS Model: 1. generating random deviates from the available probability distribution; 2. collecting statistics on the time of stage storage; 3. transferring control to other special purpose subprograms; 4. providing scheduling data for each launch. A description of each segment of subprogram HELP is shown in Table B-I. Transfer to the segments from within the HELP subprogram is accomplished by utilizing the control variable in a Fortran computed GO TO statement and selecting the proper HELP segment.

The subprograms described below are responsible for the generation of the random numbers used in the user written subprograms.

- DRAND and DRAND1 serve as two independent random number generators. The random number generation is accomplished by modulo division. The random number seeds for these two subprograms are defined by variables ISEED and ISEED1, respectively. (No support routines called.)

The following subprograms are used in determining deviates from various statistical distributions to be used by the GPSS II instructions to accurately simulate the Shuttle operations cycle.

TABLE B-I. SUBPROGRAM HELP SEGMENT DESCRIPTION

Segment	Description
1	Generates the attributes of the next launch including mission type, time between launches, Orbiter mission length, and the time between the first and second launch opportunity. (Support routine called: MISSION.)
2	Calls a special purpose subprogram that prepares information obtained from GPSS II system variables for use in the Confidence Interval Option. (Support routine called: DUMMY.)
3	Generates the deviates from a triangular distribution. (Support routine called: TRIAG1.)
4	Generates the deviates from a normal distribution. (Support routine called: RNORM.)
5	Saves the clock time when a stage has completed processing and has entered storage. (No support routine called.)
6	Computes the elapsed time a stage has spent in storage based on a first in first out basis. (No support routines called.)
7	Prepares the model for a new case study by reading the input variables and the comment card title. (No support routines called.)
8	Generates the deviates for a two-parameter exponential distribution. (Support routine called: EXPO.)
9	Generates the deviates for a lognormal distribution. (Support routine called: RLOGN.)
10	Outputs the summary table. Repeated calls are required to complete the entire table. (Support routine called: PRINT.)
11	Generates the deviates for a constant distribution. (No support routines called.)
12	Generates the deviates for a uniform distribution. (Support routine called: UNFRM1.)
13	Generates the deviates for a beta distribution. (Support routines called: BETADP and PERTXF.)

- BETADP generates the deviates for a beta distribution. (Support routines called: GAM and PERTXF.)
- DICRET produces deviates from a discrete probability distribution. This subprogram permits the model to generate random values from data that has either a $1/3$ or $1/2$ probability of selection. (Support routine called: DRAND.)
- EXPO generates the deviates for a two parameter exponential distribution. (Support routine called: DRAND1.)
- GAM generates the deviates for a gamma distribution. (Support routine called: DRAND1.)
- PERTXF converts three input parameters, namely the minimum expected time, the most likely expected time and the maximum expected time to the mean and standard deviation of a beta distribution. (No support routines called.)
- RLOGN calculates the deviates for a lognormal distribution by exponentiating the normal deviate. The mean and standard deviation of the lognormal distribution are obtained from the calling subprogram. (Support routine called: RNORM.)
- RNORM calculates the deviates for a normal distribution. The mean and standard deviation of the routine are obtained from the calling subprogram. (Support routine called: DRAND1.)
- TRIAG and TRIAG1 produce the random deviates for a triangular distribution function. These two routines are identical and independent. (Support routines called: DRAND and DRAND1, respectively.)
- UNFRM and UNFRM1 produce random deviates from a uniform distribution. The two routines are identical and independent. (Support routines called: DRAND and DRAND1, respectively.)

It should be noted that subprograms DRAND, TRIAG, and UNFRM are used exclusively by the MISSION subprogram to provide a separate and unique sequence of random numbers.

The subprogram described below is responsible for scheduling the Shuttle launches.

- MISSION is responsible for determining the parameters associated with each Shuttle launch. MISSION is used to

determine a realistic time between launches for the Shuttle system. For each launch, the mission type is defined with the possible types including Shuttle payload placement, space station logistics, tug placement and fuel supplies, Department of Defense, and others, including space rescue. The MISSION subprogram also determines the Orbiter mission length which depends on mission type, the date of the next launch, and the time between the first and second launch opportunity. (Support routines called: DICRET, TRIAG, UNFRM.)

The following subprograms are not active during the dynamic simulation of the Shuttle operations cycle, but were developed for the convenience of the user. These subprograms are used to produce the TABLE1 output which summarizes the activity time distributions defined in the System Model Description of the SOS Model.

- OUTPUT determines the type of distribution for each Shuttle processing event and outputs this information in tabular form. (Support routine called: REALN).
- REALN converts alphanumeric data into the real data used by the OUTPUT routine. (No support routines called.)
- TABLE1 scans the System Model Description for the titles of the various processing operations and for their associated time distributional parameters. (Support routine called: OUTPUT.)

The following subprograms are used when confidence intervals are requested for specific statistics generated during the dynamic simulation of the Shuttle operations cycle. Reference 3 gives detailed information on the Confidence Interval Option.

- DUMMY controls the sampling rate used in obtaining statistical observations for those statistics specified by the Confidence Interval Option. The subprogram also assures the proper collection of the statistics specified by the user. (Support routine called: SAMPLE.)
- SAMPLE makes statistical observations during the simulation on the statistics specified by the user to have confidence intervals calculated for them. At the conclusion of the simulation, a summary table is outputted that contains a descriptive heading and a corresponding confidence interval for each of the statistics specified by the user to be collected. (Support routine called: STDV.)

- STAT is used to read and store descriptive headings to be used in the output of confidence interval statistics. Each card of the System Model Description is scanned for the appropriate headings. This subprogram is not active during the actual simulation. (No support routines called.)
- STDV calculates the sample standard deviation for the statistics specified for confidence interval calculations. (No support routines called.)

The subprogram described below is responsible for a summary output of the statistics collected during the simulation of the Shuttle operations cycle.

- PRINT is a collection of various write statements which print out a summary table of various simulation parameters at the end of each computer run. The values of these parameters are transmitted to the routine through two different transfer methods. The first method transfers the values of certain GPSS II system variables (table means, facility utilization and queue times) to PRINT as arguments of the subroutine. These arguments are defined on the HELP blocks at the end of the model. The other method permits the user to retrieve the output parameters through the values stored in the GPSS II common block variables. These tables are subscripted arrays and are defined in the GPSS II reference manual. The routine is executed by a series of call statements that subsequently execute each statement of the subprogram. (Support routine called: SAMPLE.)

The GPSS II Mainline (GPSS2) routine is the only GPSS II symbolic element deck available for modification by the SOS Model user. It serves as the mainline master control for the complete GPSS II program. The GPSS2 routine is a software package available upon request from the operators of the UNIVAC 1108 EXEC VIII System. The routine is described as follows:

- GPSS2 contains the option to specify the size of the network and its associated variables by means of the Fortran parameter statements contained within the routine. The total program core size is determined by the limits of the model description as defined by the values assigned to the Fortran parameter statements. Complete instructions and other information on making changes is contained in the comments of the routine listing. (Major GPSS II routines USES, BLOCKD, SETERR, INPROC, EXECUT, PUTOFF,

ASSEMB; other support routines called: RCONWN, CSFREQ,
LETTER, STAT, and TABLE1.)

BETADP

420223*TPFS*BETADP

```

1 FUNCTION BETADP(J,SMEAN,SVAR)
2 COMMON /HEPL/ PARA(7,3),LL,ISEED,ISEED1,ISTONE(2,10),IS(2)
3 A = SMEAN
4 B = SVAR
5 X = GAM(A)
6 BETADP=X/(X+GA4(3))
7 BETADP=BETADP*(PARA(1(J,3))-PARA(1(J,2))+PARAM(J,2)
8 RETURN
9 END
    
```

```

BETA 10
BETA 20
BETA 30
BETA 40
BETA 50
BETA 60
BETA 70
BETA 80
BETA 90
    
```

9H05,P ... DICRET

9PRT,S DICRET
FURPUR HA18-06/30-13:31

```
420223*TPFS,DICRET
1 FUNCTION DICRET(JP)
2 COMMON /HEPL/PARAM(7,3),LL,ISEED,ISEED1,ISTORE(2,10),ISIZ2)
3 RNUM=DRAND(LISEED)
4 IF (PARAM(JP,2).EQ.0.160 TO 20
5 IF (RNUM.GE.0.4666160 TO 15
6 IF (RNUM.GE.0.3333160 TO 10
7 DICRET=PARAM(JP,1)
8 RETURN
9 10 DICRET=PARAM(JP,2)
10 RETURN
11 15 DICRET=PARAM(JP,3)
12 RETURN
13 20 IF (RNUM.LE.0.5160 TO 30
14 DICRET=PARAM(JP,3)
15 RETURN
16 30 DICRET=PARAM(JP,1)
17 RETURN
18 END
```

```
6HDC,P DRAND
```


DRAND		
420223*TPFS*DRAND		
1	FUNCTION DRAND(NR)	GPST2500
2	NR=NR+316231	GPST2510
3	IF(NR)5,5,6	GPST2520
4	5 NR=NR+34359738367	GPST2530
5	6 RN=NR	GPST2540
6	DRAND=RN/343597384.E2	GPST2550
7	RETURN	GPST2560
8	END	GPST2570

ENDG,P DRAND1

GPRT,S DRAND1
FURPUR HA18-06/30-13:31

DRAND1

420223*TPFS*DRAND1

1 FUNCTION DRAND1(NR1)

2 NR1=NR1*316231

3 IF(NR1)5,5,6

4 5 NR1=NR1+34359738367

5 6 RN=NR1

6 DRAND1=RN/343597384.E2

7 RETURN

8 END

@HDG,P DUMMY

@PRT,S DUMMY

FURPUR HA18-06/30-13:31

DUMMY

420223*TPFS*DUMMY

1	SUBROUTINE DUMMY(K1,K2,K3,K4,K5)	DUMY 10
2	COMMON/STAT2/NFLTS	DUMY 20
3	COMMON/STAT3/ISN	DUMY 30
4	COMMON/SEKSES/JEKS(1)	DUMY 40
5	COMMON/NODE4/JN4(1)	DUMY 50
6		DUMY 60
7	C NENTR DETERMINES THE NUMBER OF TERMINATED SIMULATIONS	DUMY 70
8	NENTR=MGT2U(JN4(276))+2	DUMY 80
9		DUMY 90
10	C DETERMINE IF 100 SIMULATIONS HAVE BEEN COMPLETED	DUMY 100
11	C IF NOT, RETURN	DUMY 110
12	C IF (NENTR.LT.100) RETURN	DUMY 120
13		DUMY 130
14	C DETERMINE IF 100 OR MORE FLIGHTS (IN INCREMENTS OF 50 FLIGHTS)	DUMY 140
15	C HAVE OCCURRED.	DUMY 150
16	C IF NOT, RETURN	DUMY 160
17	C IF (MOD(NENTR,50).NE.0) RETURN	DUMY 170
18		DUMY 180
19	C	DUMY 190
20	C	DUMY 200
21	C NFLTS DETERMINES THE NUMBER OF SIMULATIONS THAT HAVE BEEN	DUMY 210
22	C COMPLETED	DUMY 220
23	C NFLTS=JEKS(2)+JEKS(3)+JEKS(4)+JEKS(18)+JEKS(22)+JEKS(24)	DUMY 230
24	C 1 +JEKS(25)+JEKS(30)	DUMY 240
25		DUMY 250
26	C	DUMY 260
27	C DETERMINE IF THE STATISTICAL IDENTIFICATION NUMBER IS LEGAL	DUMY 270
28	C IF NOT, RETURN	DUMY 280
29	C IF (K2.LT.0.OR.K2.GT.5) RETURN	DUMY 290
30	C	DUMY 300
31	C IF OPTION 1 HAS BEEN SPECIFIED, IDENTIFY THE MAXIMUM NUMBER	DUMY 310
32	C OF STATISTICS TO BE COLLECTED	DUMY 320
33	C IF (K2.EQ.0) ISN=9	DUMY 330
34	C	DUMY 340
35	C DETERMINE THE CONFIDENCE INTERVAL IDENTIFICATION NUMBER	DUMY 350
36	C IF (K2-2)2,1,1	DUMY 360
37	C	DUMY 370
38	C	DUMY 380
39	C IF OPTION 1 HAS BEEN REQUESTED, COLLECT STATISTICS FOR THE OPERATION	DUMY 390
40	C EVALUATION PARAMETERS	DUMY 400
41	C 2 K2=K2	DUMY 410
42	C K3=K3	DUMY 420
43	C K4=K4	DUMY 430
44	C K5=K5	DUMY 440
45	C K2=1	DUMY 450
46	C K3=1	DUMY 460
47	C K4=JEKS(30)	DUMY 470
48	C K5=1	DUMY 480
49	C CALL SAMPLE(K1,K2,K3,K4,K5)	DUMY 490
50	C	DUMY 500
51	C K2=2	DUMY 510
52	C K4=NFLTS-JEKS(30)	DUMY 520
53	C CALL SAMPLE(K1,K2,K3,K4,K5)	DUMY 530
54	C	DUMY 540
55	C K2=3	DUMY 550

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56	K4=JEKS(26)	DUMY 560
57	CALL SAMPLE(K1,K2,K3,K4,K5)	DUMY 570
58	C	DUMY 580
59	K2=4	DUMY 590
60	K4=JEKS(3)	DUMY 600
61	CALL SAMPLE(K1,K2,K3,K4,K5)	DUMY 610
62	C	DUMY 620
63	K2=5	DUMY 630
64	K4=JEKS(4)	DUMY 640
65	CALL SAMPLE(K1,K2,K3,K4,K5)	DUMY 650
66	C	DUMY 660
67	K2=6	DUMY 670
68	K4=JEKS(2)	DUMY 680
69	CALL SAMPLE(K1,K2,K3,K4,K5)	DUMY 690
70	C	DUMY 700
71	K2=7	DUMY 710
72	K4=JEKS(8)	DUMY 720
73	CALL SAMPLE(K1,K2,K3,K4,K5)	DUMY 730
74	C	DUMY 740
75	K2=8	DUMY 750
76	K4=JEKS(25)	DUMY 760
77	CALL SAMPLE(K1,K2,K3,K4,K5)	DUMY 770
78	C	DUMY 780
79	K2=9	DUMY 790
80	K4=JEKS(221)+JEKS(23)+JEKS(24)	DUMY 800
81	CALL SAMPLE(K1,K2,K3,K4,K5)	DUMY 810
82	C	DUMY 820
83	C	DUMY 830
84	IF OPTION 1 HAS BEEN REQUESTED, RETURN IF(K2=EQ.0) RETURN	DUMY 840
85	C	DUMY 850
86	C	DUMY 860
87	K2=10	DUMY 870
88	K3=K3T	DUMY 880
89	K4=K4T	DUMY 890
90	K5=K5T	DUMY 900
91	CALL SAMPLE(K1,K2,K3,K4,K5)	DUMY 910
92	RETURN	DUMY 920
93	C	DUMY 930
94	C	DUMY 940
95	C	DUMY 950
96	IF OPTION 2 HAS BEEN REQUESTED, COLLECT STATISTICS FOR THE REMAINING REQUESTS	DUMY 960
97	K2=K2+9	DUMY 970
98	CALL SAMPLE(K1,K2,K3,K4,K5)	DUMY 980
99	RETURN END	DUMY 990

9H0G,P EXPO

9PRT,S EXPO
FURPUR HA18-06/30-13:31

EXPO

420223*TPFS*EXPO

```

1  FUNCTION EXPO(J)
2  COMMON /HEPL/PARAM(7,3),LL,ISEED,ISEED1,ISTORE(2,10),IS(2)
3  RNUM=DRANDI(1,ISEED1)
4  EXPO=PARAM(J,1)-PARAM(J,2)*ALOG(RNUM)
5  RETURN
6  END

```

0HUG,P GAM

0PRT,S GAM
FURPUR HA18-06/30-13:31

GAM

420223*TPFS.GAM

```

1 FUNCTION GAM(ALPHA)
2 K = ALPHA
3 FK = K
4 GAM = 0.
5 IF(K17.7,B
6 IF(K.GT.20) GO TO 30
7 PROD = 1.0
8 DO 10 I=1,K
9 PROD = PROD*DRAND1(ISEED1)
10 GAM = -ALOG(PROD)
11 7 0 = ALPHA-FK
12 IF(0-.015)12,12,11
13 11 IF(D-.985)13,20,20
14 20 W = 1.0
15 GO TO 22
16 13 A = 1.0/9
17 B = 1.0/(1.-D)
18 X = DRAND1(ISEED1)*A
19 Y = DRAND1(ISEED1)*B+X
20 IF(Y-1.)15,15,14
21 W = X/Y
22 Y = -ALOG(DRAND1(ISEED1))
23 GAM = GAM+X*Y
24 RETURN
25 30 GAM = ALPHA
26 RETURN
27 END

```

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QHDG,P GPSS2

QPT,5 GPSS2

FURPUR HA18-06/30-13:31

GPSS2

420223*1PFS.GPSS2

GPSS 11 VERSION F LEVEL 3

THE GENERAL PURPOSE SYSTEMS SIMULATOR 11 FINAL USER MANUAL IS UP-4129, AND THE FINAL GPSS 11 CODING FORM IS UP-4131. BOTH OBTAINABLE FROM HOLYOKE, MASS., VIA A SALES HELP REQUISITION THROUGH THE LOCAL UNIVAC MANAGER.

THE FOLLOWING NOTES ARE INTENDED TO SUPPLEMENT OR UPDATE INFORMATION PRESENTED IN THE GPSS USER MANUAL.....

1. EACH TIME AN OVERLAY OCCURS A WARNING TO THIS EFFECT IS PRINTED, BUT EXECUTION IS NOT INHIBITED.
2. IF JOBTAPE AND WRITE ARE USED THE NUMBER OF TRANS-ACTIONS ON TAPE MUST BE AT LEAST ONE GREATER THAN THE NUMBER REQUIRED FOR THE EXECUTION OF THE JOB.
3. EXECUTION USING XOT MAPGPS RESULTS IN A SEGMENTATION OF GPSS 11.
4. THE STANDARD UPPER LIMITS OF FIELDS GIVEN IN APPENDIX 3 OF THE GPSS USER MANUAL MAY BE CHANGED. THE METHOD FOR MAKING SUCH CHANGES HAS BEEN REVISED. SEE NOTE 7 FOR AN EXPLANATION.
5. THE EXECUTIVE COMMAND, WSETC 2, WILL CAUSE AN ASSEMBLED DECK TO BE PUNCHED OUT.
6. UPON DETECTION OF AN END COMMAND, ALL JOBTAPES WHICH WERE REFERENCED BY A WRITE BLOCK WILL HAVE AN END-OF-FILE PLACED ON THEM. REWIND OF THESE TAPES WILL NOT OCCUR.
7. THE METHOD USED TO CHANGE TABLE SIZES FOR THIS VERSION IS TO CHANGE THE VALUES GIVEN IN THE FORTAN PARAMETER STATEMENTS WHICH FOLLOW. THE TABLE TYPE REFERRED TO BY A FORTAN PARAMETER STATEMENT IS GIVEN IN THE LINE PRECEDING THAT STATEMENT.

C**BLUCKS
PARAMETER NBMAX = 400

C**FACILITIES
PARAMETER NFMAX = 15

C**STORAGES
PARAMETER NSMAX = 50

C**QUEUES
PARAMETER NQMAX = 50

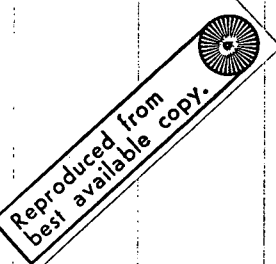
C**USER CHAINS
PARAMETER NUCMAX = 1

C**LOGIC SWITCHES
PARAMETER NLSMAX = 25

C**SAVEV LOCATIONS
PARAMETER NSLMAX = 50

USCH

USCH



56 C
57 C**FUNCTIONS
58 PARAMETER NFMNAX = 10
59 C
60 C**TABLES AND QTABLES (COMBINED TOTAL)
61 PARAMETER NJRMAX = 30
62 C
63 C**VARIABLE STATEMENTS
64 PARAMETER NVMAX = 30
65 C
66 C**COMMON CORE AREA
67 PARAMETER NCCMAX = 4500
68 C
69 C**TRANSACTIONS (MAXIMUM ALLOWED IN SYSTEM AT ANY GIVEN TIME)
70 PARAMETER NTMAX = 100
71 C
72 C**PARAMETERS PER TRANSACTION (MUST BE LESS THAN OR EQUAL TO 30)
73 PARAMETER NPRMAX=20
74 C*****THE FOLLOWING CARDS ARE NOT TO BE DELETED, NOR CHANGED
75 C UNLESS SO NOTED IN COMMENT CARDS.*****
76 PARAMETER NPRM=((NPRMAX+1)/2)*2 ,
77 A NPX2 = (NPRM/2-NPRM/4-NPRM/6-NPRM/10-NPRM/12
78 B -NPRM/14-NPRM/20-NPRM/22-NPRM/26
79 C +NPRM/28-NPRM/30 ,
80 D NPX4 = NPRM/4-NPRM/8-NPRM/12-NPRM/20-NPRM/24-NPRM/28 ,
81 E NPX6 = NPRM/6-NPRM/12-NPRM/18-NPRM/30 ,
82 F NPX8 = NPRM/8-NPRM/16-NPRM/24 ,
83 G NPX10 = NPRM/10-NPRM/20-NPRM/30 ,
84 H NP2 = NPX2*NTMAX +1 -NPX2 ,
85 I NP4 = NPX4*NTMAX +1 -NPX4 ,
86 J NP6 = NPX6*NTMAX +1 -NPX6 ,
87 PARAMETER NP8 =NPX8*NTMAX +1 -NPX8 ,
88 1 NP10 = NPX10*NTMAX +1 -NPX10 ,
89 2 NP12 = (NPRM/12-NPRM/24)*NTMAX +1 -(NPRM/12-NPRM/24),
90 3 NP14 = (NPRM/14-NPRM/28)*NTMAX +1 -(NPRM/14-NPRM/28),
91 4 NP16 = (NPRM/16)*NTMAX +1 -NPRM/16 ,
92 5 NP18 = (NPRM/18)*NTMAX +1 -NPRM/18 ,
93 6 NP20 = (NPRM/20)*NTMAX +1 -NPRM/20 ,
94 7 NP22 = (NPRM/22)*NTMAX +1 -NPRM/22 ,
95 8 NP24 = (NPRM/24)*NTMAX +1 -NPRM/24 ,
96 9 NP26 = (NPRM/26)*NTMAX +1 -NPRM/26 ,
97 A NP28 = (NPRM/28)*NTMAX +1 -NPRM/28 ,
98 B NP30 = (NPRM/30)*NTMAX +1 -NPRM/30
99 COMMON/NODE1/JN1(NBMAX)
100 COMMON/NODE2/JN2(NBMAX)
101 COMMON/NODE3/JN3(NBMAX)
102 COMMON/NODE4/JN4(NBMAX)
103 COMMON/NODE5/JN5(NBMAX)
104 COMMON/EQ1/JF1(NFMAX)
105 COMMON/EQ2/JF2(NFMAX)
106 COMMON/EQ3/JF3(NFMAX)
107 COMMON/EQ4/JF4(NFMAX)
108 COMMON/EQ5/JF5(NFMAX)
109 COMMON/EQ6/JF6(NFMAX)
110 COMMON/EQ7/JF7(NFMAX)
111 COMMON/STOR1/JS1(NSMAX)

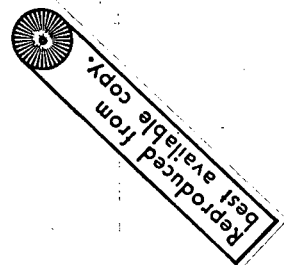
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112	COMMON/STOR2/JS2(NSMAX)	
113	COMMON/STOR3/JS3(NSMAX)	
114	COMMON/STOR4/JS4(NSMAX)	
115	COMMON/STOR5/JS5(NSMAX)	
116	COMMON/STOR6/JS6(NSMAX)	
117	COMMON/STOR7/JS7(NSMAX)	
118	COMMON/QUE1/JQ1(NQMAX)	
119	COMMON/QUE2/JQ2(NQMAX)	
120	COMMON/QUE3/JQ3(NQMAX)	
121	COMMON/QUE4/JQ4(NQMAX)	
122	COMMON/QUE5/JQ5(NQMAX)	
123	COMMON/QUE6/JQ6(NQMAX)	
124	COMMON/LOG1X/JL1(NLSMAX)	
125	COMMON/EKSES/JEKS(NSLMAX)	
126	COMMON/FN1/JYLQCS(NFNMAX)	
127	COMMON/FEN2/JXLQCS(NFNMAX)	
128	COMMON/FN3/JSLQCS(NFNMAX)	
129	COMMON/TAB1/JTLQCS(NTGMAX)	
130	COMMON/TAB2/JTMODE(NTGMAX)	
131	COMMON/TAB3/JLOWRS(NTGMAX)	
132	COMMON/TAB4/JTINCS(NTGMAX)	
133	COMMON/TAB5/JTLASI(NTGMAX)	
134	COMMON/TAB6/JTLNUM(NTGMAX)	
135	COMMON/TAB7/JTLARG(NTGMAX)	
136	COMMON/TAB8/TSQR(NTGMAX)	
137	COMMON/TAB9/TWARG(NTGMAX)	
138	COMMON/TAB10/TNSQR(NTGMAX)	
139	COMMON/VARS/JVLQCS(NVFX)	
140	COMMON/WORDS/JWORDS(NCCMAX)	
141	COMMON/UC1/JU1(NUCMA)	USCH
142	COMMON/UC2/JU2(NUCMA)	USCH
143	COMMON/UC3/JU3(NUCMA)	USCH
144	COMMON/UC4/JU4(NUCMA)	USCH
145	COMMON/UC5/JU5(NUCMA)	USCH
146	COMMON/TRAN00/JTSTAT(NTMAX)	
147	COMMON/TRAN1/JNDI(NTMAX)	
148	COMMON/TRAN2/JCHAT(NTMAX)	
149	COMMON/TRAN3/JMOVE(NTMAX)	
150	COMMON/TRAN4/JNNWD(NTMAX)	
151	COMMON/TRAN5/JC1(NTMAX)	
152	COMMON/TRAN6/JC2(NP2)	
153	COMMON/TRAN7/JC3(NP4)	
154	COMMON/TRAN8/JC4(NP6)	
155	COMMON/TRAN9/JC5(NP8)	
156	COMMON/TRAN10/JC6(NP10)	
157	COMMON/TRAN12/JC7(NP12)	
158	COMMON/TRAN14/JC8(NP14)	
159	COMMON/TRAN16/JC9(NP16)	
160	COMMON/TRAN18/JC10(NP18)	
161	COMMON/TRAN20/JC11(NP20)	
162	COMMON/TRAN22/JC12(NP22)	
163	COMMON/TRAN24/JC13(NP24)	
164	COMMON/TRAN26/JC14(NP26)	
165	COMMON/TRAN28/JC15(NP28)	
166	COMMON/TRAN30/JC16(NP30)	
167	COMMON K(100)	

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168 COMMON LPRI(128),LPRIOR(128)
169 COMMON ICHAR(70),KTYPE(43),KGATE(12),KCONTR(7),KSV(50),
170 1 KCOMP(6),KSELEC(7),LX(6)
171 COMMON/STAT3/ISN
172 COMMON/STATS/NHEAD
173 DIMENSION NHEAD(14,11)
174 DIMENSION FWORDS(11)
175 EQUIVALENCE(JWORDS(1),FWORDS(1))
176 EQUIVALENCE(K(1),KASYN1),K(2),KASYM2),K(3),KNODES),K(4),KEARS),
177 1 (K(5),KSTORS),K(6),KQUES),K(7),KVAR5),K(8),KLOGIX),
178 2 (K(9),KEKSES),K(10),KFNS),K(11),KTAUS),K(12),KWORDS),
179 3 (K(13),KTRANS),K(14),KBRAD),K(15),KASMBL),K(16),KIT),
180 4 (K(17),KOT),K(18),KUSERS)
181 EQUIVALENCE(K(79),KPARAM),K(71),INDFLO),K(72),INDEND),
182 1 (K(55),IFATAL)
183
184 C
185 C*****THE FOLLOWING NOTES PROVIDES TO THE USER OF THE MSFC-SUS MODEL
186 THE NECESSARY INFORMATION CONCERNING @SETC COMMAND CONTROL OPTIONS.
187 C 1. A @SETC 3 COMMAND PLACED BEFORE THE @XQT CONTROL CARD WILL
188 ELIMINATE THE PRINTOUT OF THE MODEL LISTING. ONLY SUMMARY
189 RESULTS AND STANDARD GPSS OUTPUT WILL BE PRINTED.
190 C 2. A @SETC 4 COMMAND PLACED BEFORE THE @XQT CONTROL CARD WILL
191 GENERATE A TABLE CONTAINING THE DISTRIBUTION TIME PARAMETERS
192 FOR THE DIFFERENT PROCESSING OPERATIONS.
193 C NOTE: THAT THE MODEL MUST BE ADDED TO THE RUN STREAM IN ORDER
194 TO PRODUCE THIS OUTPUT TABLE.
195 C @SETC 4
196 @XQT GPSSSEB.MAPGPS
197 @ADDU MODEL.
198 @ADDU MODEL.
199 COMMENT CARD FOR CASE 1
200 DATA CARD 1
201 END
202
203 C 3. A @SETC 5 COMMAND PLACED BEFORE THE @XQT CONTROL CARD WILL
204 DETERMINE THE HEADINGS TO BE USED IN THE CONFIDENCE INTERVAL
205 STATISTICS COLLECTED BY THE MODEL. THIS COMMAND IS USED ONLY
206 WHEN CONFIDENCE INTERVAL STATISTICS ARE DESIRED.
207 C NOTE: TO COLLECT THE HEADINGS TO BE ASSOCIATED WITH THE
208 STATISTICS COLLECTED, THE MODEL MUST BE ADDED TO THE RUN STREAM.
209 C @SETC 5
210 @XQT GPSSSEB.MAPGPS
211 @ADDU MODEL.
212 @ADDU MODEL.
213 COMMENT CARD FOR CASE 1
214 DATA CARD 1
215 END
216
217 DIMENSION ICTLCD(4),JCTLCD(4),KCTLCD(4)
218 DATA ICTLCD/'RASSG.T DUMMY.F',
219 1 JCTLCD/'@BRKPT PRINTS/DUMMY',
220 2 KCTLCD/'@BRKPT PRINTS',
221 CALL RCONDD(MPRINT)
222 NPRINT=MGETT2(MPRINT)
223 IF(MPRINT.EQ.4)CALL TABLE1
224 IF(MPRINT.EQ.5) CALL STAT
225 IF(MPRINT.NE.3)GO TO 5

```



```

224 CALL CSFREQ(JCTLCN)
225 CALL CSFREQ(JCTLCD)
226 5 CALL USES
227 KPARAM = NPRM
228 KIT = 5
229 KOT = 6
230 KRAND = 1220703125
231 KNODES = HBMAX
232 KEQS = NFMAX
233 KSTORS = NSMAX
234 KQUES = NOMAX
235 KVAR5 = NYMAX
236 KLOGIX = NLSMAX
237 KEKSES = NSLMAX
238 KFNS = NFMAX
239 KIABS = NTQMAX
240 KWORDS = NCCMAX
241 KTRANS = NIMAX
242 KUSERS = NUCMAX
243 CALL BLOCKD
244 CALL SETERR
245 10 CALL IMPROC($20,$30)
246 20 IF (INDFLO.NE.O) CALL FLOW
247 C 10 PERMIT EXECUTION OF A JOB WITH ILLEGAL INPUT CARDS, OR IO
C 20 PERMIT EXECUTION FOLLOWING A RESET OR CLEAR CARD WHERE A GPSS II
C ERROR HAS ALREADY OCCURRED, REMOVE THE FOLLOWING CARD
248 C IF (IFATAL.NE.O) GO TO 10
249 C IF (IFATAL.NE.O) GO TO 10
250 C IF (INDEND.NE.O) GO TO 10
251 C IF (INDEND.NE.O) GO TO 10
252 C IF (INPRINT.NE.O) GO TO 25
253 CALL CSFREQ(JCTLCD)
254 CALL LETTER
255 25 CALL EXECUT
256 CALL PUTOUT
257 GO TO 10
258 30 CALL ASSEMB
259 GO TO 10
260 END

```

USCH

B-19

GRUG,P HELP

GPRT,S HELP
FURPUR HA18-06/30-13:31

HELP

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420723*TPFS,HELP
1      SUBROUTINE HELP(K1,K2,K3,K4,K5)      HELP 10
2      COMMON /HEPL/PARAM(7,3),LL,ISEED,ISEED1,ISTORE(2,10),IS(2)      HELP 20
3      COMMON /HLPRAT/NBOOS,NORHT,NLUTS,IITLE(20),NSEED,NSEED1
4      COMMON/ISTAT/NCONF
5      25 GO TO (1,2,3,4,5,6,7,8,9,10,11,12,13),K1
6      C
7      C*****GENERATE REQUIREMENTS FOR NEXT LAUNCH
8      C
9      1 CONTINUE
10     CALL MISSION(NP1,NP2,NP3,NP4,NP5,MODEL)
11     K2=NP1      HELP 160
12     K3=NP2      HELP 170
13     K4=NP3      HELP 180
14     K5=NP5
15     RETURN
16     2 CONTINUE
17     C
18     C*****COLLECT STATISTICAL DATA FOR CONFIDENCE INTERVALS
19     C
20     CALL DUMMY(K1,K2,K3,K4,K5)
21     RETURN
22     C
23     C*****GENERATE DEVIATE FOR TRIANGULAR DISTRIBUTION
24     C
25     3 PARAM(1,1)=K2      HELP 260
26     PARAM(1,2)=K3      HELP 270
27     PARAM(1,3)=K4      HELP 280
28     K5=TRIANG(1)
29     RETURN
30     C
31     C*****GENERATE DEVIATE FOR NORMAL DISTRIBUTION
32     C
33     4 PARAM(1,1)=K2      HELP 340
34     PARAM(1,2)=K3      HELP 350
35     K4=RNORM(1)
36     K5=K4
37     RETURN
38     C
39     C*****STORE THE TIME THAT THE BOOSTER (K2=1) OR ORBITER (K2=2) WAS
40     C*****PUT IN STORAGE.
41     C
42     5 IS(K2)=IS(K2)+1      HELP 440
43     K=IS(K2)      HELP 450
44     ISTORE(K2,K)=K3      HELP 470
45     RETURN
46     C
47     C*****COMPUTE THE TIME THAT THE BOOSTER OR ORBITER WAS PUT IN STORAGE
48     C*****BASED ON FIRST-IN-FIRST-OUT.
49     6 K4=K3-ISTORE(K2,1)      HELP 500
50     IS(K2)=IS(K2)-1      HELP 520
51     K=IS(K2)      HELP 530
52     IF (K.EQ.0) RETURN      HELP 540
53     DO 20 I=1,K
54     20 ISTORE(K2,I)=ISTORE(K2,I+1)      HELP 560
55     RETURN      HELP 570

```

B-20

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56 7 CONTINUE HELP 580
57 C HELP 590
58 C*****INITIALIZE MISSION SUBROUTINE AND PREPARE FOR NEW CASE HELP 600
59 C HELP 610
60 LL=0 HELP 620
61 C HELP 650
62 C*****READ CASE STUDY COMMENT CARD HELP 670
63 C HELP 700
64 READ(5,101)(TITLE(I),I=1,20) HELP 710
65 101 FORMAT(20A4) HELP 720
66 C
67 C*****READ THE VERSION CODE OF THE TRAFFIC MODEL TO BE SIMULATED
68 C*****READ IN THE NUMBER OF BOOSTERS, ORBITERS & LUTS
69 C*****READ IN THE LEVEL OF SIGNIFICANCE
70 C*****READ IN THE RANDOM NUMBER SEED FOR MISSION MODEL PROGRAM(1SEED)
71 C*****READ IN THE RANDOM NUMBER SEED FOR DISTRIBUTION GENERATORS(1SEED1) HELP 740
72 C
73 READ(5,103)MODEL,NBOOS,NORBI,NLUTS,NCONF,1SEED,1SEED1
74 103 FORMAT(I1,I4,3I5,40X,2I10)
75 1S(1)=NBOOS HELP 770
76 1S(2)=NORBI HELP 780
77 K2=NBOOS HELP 790
78 K3=NORBI HELP 800
79 K4=NLUTS
80 NSEED=1SEED
81 NSEED1=1SEED1
82 RETURN HELP 810
83 8 CONTINUE HELP 820
84 C HELP 830
85 C*****GENERATE DEViate FOR A TWO PARAMETER EXPONENTIAL DISTRIBUTION HELP 840
86 C HELP 850
87 PARAM(1,1)=K2 HELP 860
88 NBETA=K3-K2
89 PARAM(1,2)=NBETA
90 K5=EXP0(1)
91 RETURN
92 9 CONTINUE
93 C
94 C*****GENERATE DEViate FOR THE LOG NORMAL DISTRIBUTION
95 C
96 F2=FLOAT(K2)/1000.
97 F3=FLOAT(K3)/1000.
98 PARAM(1,2)=SQRT(ALOG(F3**2/F2**2+1.))
99 PARAM(1,1)=ALOG(F2)-.5*(PARAM(1,2))**2
100 K5=RLOGN(1)*1000
101 RETURN
102 10 CONTINUE
103 C
104 C*****PRINT OUT OF SUMMARY TABLE
105 C
106 NCALL=NCALL+1
107 CALL PRINT(NCALL,K2,K3,K4,K5)
108 RETURN
109 11 CONTINUE
110 C
111 C*****THIS FUNCTION PERMITS THE ADVANCE BLOCK TO BE INCREMENTED BY A

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HELP

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112 C*****CONSTANT
113 C*****TRANSFER THE VALUE OF ARGUMENT K2 TO ARGUMENT K5.
114 C
115 C GPSS OUTPUT CAN BE OBTAINED FROM EITHER THE K4 OR K5 FIELDS
116 K5=K2
117 K4=K2
118 RETURN
119 12 CONTINUE
120 C
121 C*****GENERATE DEViate FOR UNIFORM DISTRIBUTION
122 C
123 A = K2
124 B = K3
125 C GPSS OUTPUT CAN BE OBTAINED FROM EITHER THE K4 OR K5 FIELDS
126 K4 = UNFRM1(A,B)
127 K5=K4
128 RETURN
129 13 CONTINUE
130 C
131 C*****GENERATE DEViate FOR BETA DISTRIBUTION
132 C
133 PARAM(1,1) = K3
134 PARAM(1,2) = K2
135 PARAM(1,3) = K4
136 CALL PERTXF(1,SMEAN,SVAR)
137 K5 = BETADP(1,SMEAN,SVAR)
138 RETURN
139 END

```

HELP1070

BHDG,P MAP

GPRT,S MAP

FURPUR HA18-06/30-13:31

MISSION

420223*TPFS*MISSION

1 SUBROUTINE MISSION(MP1,MP2,MP3,MP4,NP5,K)

2 C

3 C*****THIS PROGRAM GENERATES TRAFFIC REQUIREMENTS FOR THE SHUTTLE

4 OPERATIONAL MISSIONS.

5 C*****THE SELECTION OF NEXT SCHEDULED LAUNCH IS BASED ON THE FREQUENCY

6 OF MISSION OCCURRENCE.

7 C

8 REAL LATIME

9 DIMENSION OPPORT(5,3),MILGTH(5,3)

10 DIMENSION LCONS(5),LOCATE(5),INTERL(5),LFREQ(5,5)

11 COMMON /HEFL/PARAM(7,3),LL,ISEED,ISEED1,ISTORE(2,10),IS(2)

12 COMMON /NISPT/HELTAI,NELTA2,NELTA3,NELTA4,NELTAS

13 C

14 C*****VARIABLE K DEFINES THE TRAFFIC MODEL TO BE SIMULATED

15 K=1 - 20 LAUNCHES/YEAR

16 K=2 - 35 LAUNCHES/YEAR

17 K=3 - 45 LAUNCHES/YEAR

18 K=4 - 55 LAUNCHES/YEAR

19 K=5 - 75 LAUNCHES/YEAR

20 C*****ARRAY LFREQ CONTAINS THE FREQUENCY REQUIREMENTS FOR THE 5 TRAFFIC

21 MODELS. EACH ROW DEFINES A NEW MODEL.(FLIGHTS/YEAR)

22 DATA LFREQ/10,8,0,0,2,

23 A 14,6,2,12,1,

24 B 16,8,4,16,1,

25 C 20,8,8,19,2,

26 D 35,8,6,19,5/

27 C*****DATA ON THE MINIMUM TIME BETWEEN LAUNCH REQUESTS(DAYS)

28 DATA LCONS/5,3,3,2,1/

29 C*****DATA FOR THE TIME UNTIL SECOND LAUNCH OPPORTUNITY(HOURS)

30 DATA OPRT/8,3,6,3,8,3,6,3,3,0,

31 A 14,17,0,0,14,17,8,33,

32 B 24,0,14,17,24,0,14,17/

33 C*****DATA ON THE RANGE OF TYPICAL ORBITER MISSION LENGTHS(DAYS)

34 DATA MILGTH/2,1,4,1,1,

35 A 3,2,5,3,5,

36 B 5,3,6,5,20/

37 C*****VARIABLE NTYPE IS THE NUMBER OF DIFFERENT TYPES OF MISSIONS

38 NTYPE=5

39 C*****VARIABLE I DEFINES THE MISSION TYPE

40 I=1 - SHUTTLE PAYLOAD PLACEMENT

41 I=2 - SPACE STATION LOGISTICS

42 I=3 - TUG P/L PLACEMENT & FUEL RESUPPLY

43 I=4 - DEPARTMENT OF DEFENSE

44 I=5 - SPACE RESCUE & OTHER TYPES

45 C

46 C

47 C*****GO THRU THIS LEG ON THE FIRST CALL ONLY*****

48 C

49 IFILL,NE,0160 TO 2

50 C*****VARIABLE LL IS SET = 1 WHEN EXECUTION PASSES THRU THIS LEG

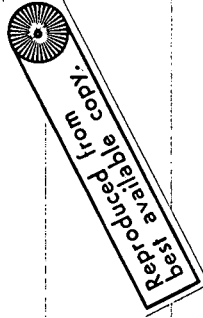
51 LL = 1

52 SAVE = 0.0

53 C*****DETERMINE THE FIRST LAUNCH DATE FOR EACH OF THE MISSION TYPES

54 C*****EVENLY SPACED LAUNCH DATES HAVE BEEN ASSUMED FOR THE FIRST 5

55 LAUNCHES.



NISSON

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56 LR(LFREQ(I,K)+LFREQ(2,K)+LFREQ(3,K)+LFREQ(4,K)+LFREQ(5,K))
57 LFIRST=260/LR
58 DO 7 I=1,NTYPE
59 LDATE(I)=LDATE(I)+LDATE(1)
60 C*****IF MISSION TYPE IS NOT DEFINED, SET THE LAUNCH DATE = 99999999
61 IF(LFREQ(I,K).EQ.0) LDATE(I)=99999999
62 C*****MISSIONS ARE TO BE SCHEDULED AT THE SAME FREQUENCY WITH 10 PERCENT
63 C DEVIATION
64 INTERL(I)=50*FLOAT(260/LFREQ(I,K))
65 IF(LFREQ(I).EQ.0) INTERL(I)=INTERL(1)+2
66 7 CONTINUE
67 C*****
68 2 CONTINUE
69 C
70 C*****CALCULATE THE NEXT LAUNCH TRAFFIC REQUIREMENT
71 C
72 NMN=MINO(LDATE(1),LDATE(2),LDATE(3),LDATE(4),LDATE(5))
73 DO 20 I=1,NTYPE
74 IF(LDATE(I).EQ.NMN) GO TO 21
75 20 CONTINUE
76 MERROR=1
77 WRITE(6,100)MERROR
78 100 FORMAT(1X,6HERROR ,11)
79 C
80 C*****ASSIGN LAUNCH HOUR FOR MISSION ASSIGNMENT
81 C
82 21 NP2=I
83 TIME=LDATE(I)
84 HOUR = UNFM(0.,24.)
85 LATIME = TIME + HOUR/24.
86 NP3=LATIME*1000.-SAVE
87 NP3=IABS(NP3)
88 C
89 C*****GENERATE THE HISTOGRAM FOR THE TIME BETWEEN LAUNCH REQUESTS
90 C
91 IF(NP3.LT.2000)NLTAL=NLTAL+1
92 IF(NP3.GE.2000.AND.NP3.LT.3000)NLTAL2=NLTAL2+1
93 IF(NP3.GE.3000.AND.NP3.LT.5000)NLTAL3=NLTAL3+1
94 IF(NP3.GE.5000.AND.NP3.LT.10000)NLTAL4=NLTAL4+1
95 IF(NP3.GE.10000.AND.NP3.LT.20000)NLTAL5=NLTAL5+1
96 SAVE=LATIME*1000.
97 NP4=SAVE
98 C
99 C*****CALCULATE THE FIRST LAUNCH OPPORTUNITY AFTER THE NOMINAL LAUNCH TIME
100 C
101 DO 40 J=1,3
102 PARAM(I,J)=OPPORT(I,J)
103 40 CONTINUE
104 XOPPOR = DCRET(I)
105 NPS=XOPPOR/24.*1000.
106 C
107 C*****SELECT THE MISSION LENGTH OF THE ORBITER
108 C
109 DO 50 J=1,3
110 PARAM(I,J)=MILGTH(I,J)
111 50 CONTINUE

```



```

112 NP1=TRIAG(I)*1000.
113 C
114 C*****CALCULATE THE DATE OF THE NEXT LAUNCH
115 C
116 NCOUNT=0
117 DO CONTINUE
118 NCOUNT=NCOUNT+1
119 A=INTERL(I)
120 B=INTERL(I)
121 NVAR=UNFRM(A,B)
122 NDAY=LDATE(I)+240/LFREQ(I,K)*NVAR
123 DO 65 J=1,N1YPE
124 C
125 C*****AFTER 50 RESCHEDULES , THE CONSTRAINT MAY BE OVERRIDDEN
126 C
127 IF(NCOUNT.EQ.50)WRITE(6,107)
128 107 FORMAT(///' *TRAFFIC MODEL VIOLATED MINIMUM TIME BETWEEN LAUNCH
129 *REQUESTS CONSTRAINT. TYPE=,I3)
130 IF(NCOUNT.EQ.50)GO TO 6A
131 C
132 C*****DETERMINE IF TIME BETWEEN LAUNCH REQUESTS MEETS THE MINIMUM REQ.
133 C IF NOT- TRY TO SCHEDULE ANOTHER LAUNCH DATE
134 C
135 NTEST=NDAY-LDATE(J)
136 IF(ABS(NTEST).LE.LCONST(K))GO TO 60
137 65 CONTINUE
138 68 CONTINUE
139 LDATE(I)=NDAY
140 RETURN
141 END

```

B-25

PHDG,P OUTPUT

OPRT,S OUTPUT
FUNKP HA18-06/30-13:31

OUTPUT

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420223*TPFS.OUTPUT
1 SUBROUTINE OUTPUT (ICOM,LOC,NAME,IX,IY,IZ,ISELM,NBA,NBB,MEAN,MOD)
2 C
3 C SUBROUTINE TO DETERMINE TYPE OF DISTRIBUTION AND WRITE
4 C CORRESPONDING DATA.
5 C
6 DIMENSION NAME(2),ICOM(11)
7 DATA K9/2HK9,K3/2HK3,K4/2HK4,K8/2HK8/
8 DATA K11/3HK11,K12/3HK12,K13/3HK13/
9 C
10 C TEST TO DETERMINE IF DISTRIBUTION IS LOGNORMAL (K9)
11 IF (IX.EQ.K9) GO TO 1
12 C
13 C TEST TO DETERMINE IF DISTRIBUTION IS TRIANGULAR (K3)
14 IF (IX.EQ.K3) GO TO 2
15 C
16 C TEST TO DETERMINE IF DISTRIBUTION IS NORMAL (K4)
17 IF (IX.EQ.K4) GO TO 3
18 C
19 C TEST TO DETERMINE IF DISTRIBUTION IS EXPONENTIAL (K8)
20 IF (IX.EQ.K8) GO TO 4
21 C
22 C TEST TO DETERMINE IF DISTRIBUTION IS CONSTANT (K11)
23 IF (IX.EQ.K11) GO TO 5
24 C
25 C TEST TO DETERMINE IF DISTRIBUTION IS UNIFORM (K12)
26 IF (IX.EQ.K12) GO TO 6
27 C
28 C TEST TO DETERMINE IF DISTRIBUTION IS BETA (K13)
29 IF (IX.EQ.K13) GO TO 7
30 RETURN
31 I CONTINUE
32 C
33 C CALL SUBROUTINE REALM TO CONVERT ALPHANUMERIC NUMBERS TO
34 C REAL NUMBERS.
35 CALL REALM (RES1,IY)
36 CALL REALM (RES2,IZ)
37 C
38 RES1 = RES1/1000.0
39 RES2 = RES2/1000.0
40 C
41 C WRITE RESULTS FOR LOGNORMAL DISTRIBUTION
42 WRITE (6,90) ICOM,RES1,RES2
43 WRITE (6,84)
44 RETURN
45
46
47 C
48 C CALL SUBROUTINE REALM TO CONVERT ALPHANUMERIC NUMBERS TO
49 C REAL NUMBERS.
50 CALL REALM (RES1,IY)
51 CALL REALM (RES2,IZ)
52 CALL REALM (RES3,MEAN)
53 C
54 RES1 = RES1/1000.0
55 RES2 = RES2/1000.0
56 RES3 = RES3/1000.0

```

OUTPUT

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56 C
57 C WRITE RESULTS FOR TRIANGULAR DISTRIBUTION
58 WRITE (6,91) ICOM,RES1,RES2,RES3
59 WRITE(6,84)
60 RETURN
61 3 CONTINUE
62 C
63 C CALL SUBROUTINE REALN TO CONVERT ALPHANUMERIC NUMBERS TO
64 C REAL NUMBERS.
65 CALL REALN (RES1,IY)
66 CALL REALN (RES2,IZ)
67 C
68 RES1 = RES1/1000.0
69 RES2 = RES2/1000.0
70 C
71 C WRITE RESULTS FOR NORMAL DISTRIBUTION
72 WRITE (6,92) ICOM,RES1,RES2
73 WRITE(6,84)
74 RETURN
75 4 CONTINUE
76 C
77 C CALL SUBROUTINE REALN TO CONVERT ALPHANUMERIC NUMBERS TO
78 C REAL NUMBERS.
79 CALL REALN (RES1,IY)
80 CALL REALN (RES2,IZ)
81 C
82 RES1 = RES1/1000.0
83 RES2 = RES2/1000.0
84 C
85 C WRITE RESULTS FOR EXPONENTIAL DISTRIBUTION
86 WRITE (6,93) ICOM,RES1,RES2
87 WRITE(6,84)
88 RETURN
89 5 CONTINUE
90 C
91 C CALL SUBROUTINE REALN TO CONVERT ALPHANUMERIC NUMBERS TO
92 C REAL NUMBERS
93 CALL REALN (RES3,IY)
94 C
95 RES3=RES3/1000.0
96 C
97 C WRITE RESULTS FOR CONSTANT DISTRIBUTION
98 WRITE (6,94)ICOM,RES3
99 WRITE (6,84)
100 RETURN
101 6 CONTINUE
102 C
103 C CALL SUBROUTINE REALN TO CONVERT ALPHANUMERIC NUMBERS TO
104 C REAL NUMBERS
105 CALL REALN(RES1,IY)
106 CALL REALN(RES2,IZ)
107 RES3=(RES1+RES2)/2000.0
108 RES1=RES1/1000.0
109 RES2=RES2/1000.0
110 C
111 C WRITE RESULTS FOR UNIFORM DISTRIBUTION

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OUTPUT

```
112 WRITE(6,95) ICOM,RES1,RES3,RES2
113 WRITE (6,84)
114 RETURN
115 7 CONTINUE
116 C
117 C CALL SUBROUTINE REALN TO CONVERT ALPHANUMERIC NUMBERS TO
118 REAL NUMBERS
119 CALL REALN(RES1,IY)
120 CALL REALN(RES2,IZ)
121 CALL REALN(RES3,MEAN)
122 RES1=RES1/1000.0
123 RES2=RES2/1000.0
124 RES3=RES3/1000.0
125 C
126 C WRITE RESULTS FOR BETA DISTRIBUTION
127 WRITE (6,96) ICOM,RES1,RES2,RES3
128 WRITE (6,84)
129 RETURN
130 C
131 C 84 FORMAT (IX,IHI,67X,IHI,13X,IHI,14X,IHI,17X,IHI,13X,IHI)
132 C
133 90 FORMAT(IX,IHI,IX,A5,2A6,8A6,IX,IHI,IX,IHLOGNORMAL
134 1 IX,IHI,IX,5HMEAN=F7.4,IX,IHI,IX,8HSTD.DEV =,F7.4,IX,IHI,13X,IHI)
135 C
136 91 FORMAT(IX,IHI,IX,A5,2A6,8A6,IX,IHI,IX,IHTRIANGULAR,
137 1 IX,IHI,IX,5HMIN =,F7.4,IX,IHI,IX,8HMODE =,F7.4,
138 2 IX,IHI,IX,4HMAX=,F7.4,IX,IHI)
139 C
140 92 FORMAT(IX,IHI,IX,A5,2A6,8A6,IX,IHI,IX,IHNORMAL
141 1 IX,IHI,IX,5HMEAN=F7.4,IX,IHI,IX,8HSTD.DEV =,F7.4,IX,IHI,13X,IHI)
142 C
143 93 FORMAT(IX,IHI,IX,A5,2A6,8A6,IX,IHI,IX,IHEXPONENTIAL,
144 1 IX,IHI,IX,5HMIN =,F7.4,IX,IHI,IX,8HMEAN =,F7.4,IX,IHI,13X,IHI)
145 C
146 94 FORMAT(IX,IHI,IX,A5,2A6,8A6,IX,IHI,IX,IHCONSTANT,
147 1 IX,IHI,IX,5HMEAN=,F7.4,IX,IHI,17X,IHI,13X,IHI)
148 95 FORMAT(IX,IHI,IX,A5,2A6,8A6,IX,IHI,IX,IHUNIFORM
149 1 IX,IHI,IX,5HMIN =,F7.4,IX,IHI,IX,8HMEAN =,F7.4,
150 2 IX,IHI,IX,4HMAX=,F7.4,IX,IHI)
151 C
152 96 FORMAT(IX,IHI,IX,A5,2A6,8A6,IX,IHI,IX,IHBETA
153 1 IX,IHI,IX,5HMIN =,F7.4,IX,IHI,IX,8HMEAN =,F7.4,
154 2 IX,IHI,IX,4HMAX=,F7.4,IX,IHI)
155 END
```

ENDG,P PERTXF

ENDP,S PERTXF

FURPUR HA18-06/30-13:31

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420223*TPFS,PERTXF
1 SUBROUTINE PERTXF(J,SMEAN,SVAR)
2 COMMON /HEPL/ PARAM(7,3) IL,ISEED,ISTORE(2,10),IS(2)
3 SVAR = (PARAM(J,3)-PARAM(J,2))*2./36.
4 SMEAN = (PARAM(J,1)*4. + PARAM(J,2) + PARAM(J,3))/6.*0
5 C
6 C*** CALCULATE THE MEAN AND VARIANCE FOR THE GIVEN PARAMETERS
7 C***
8 C***
9 C***
10 C***
11 C***
12 C***
13 C***
14 C
15 BMEAN = (SMEAN - PARAM(J,2))/(PARAM(J,3)-PARAM(J,2))
16 BVAR = SVAR/(PARAM(J,3)-PARAM(J,2))*2.*0
17 SMEAN = BMEAN*(BMEAN+1.0-BMEAN)/BVAR-1.0
18 SVAR = SMEAN*(1.0-BMEAN)/BMEAN
19 RETURN
20 END

```

CHUG,P PRINT

9PRT,S PRINT
FURPUR HA18-06/30-13:31

```

420223*TPFS.PRINT
1 SUBROUTINE PRINT(INCALL,K2,K3,K4,K5)
2 C
3 C*****THIS ROUTINE PRINTS OUT THE SUMMARY RESULTS TABLE*****
4 C
5 REAL MAHUF,LUTREE
6 INTEGER P
7 INTEGER GLOB
8 COMMON /HLPRT/NB00S,NORBI,NLUTS,TITLE(20),NSEED,NSEEDI
9 COMMON /MISPRT/NELTA1,NELTA2,NELTA3,NELTA4,NELTAS
10 COMMON/EQ2/JF2(1)
11 COMMON/SIOR/JSI(1)
12 COMMON/STOR2/JS2(1)
13 COMMON/QUES/JQ5(1)
14 COMMON/QUE6/JQ6(1)
15 COMMON/EKSES/JEKS(1)
16 COMMON/WAY/P
17 COMMON/STAT3/ISN
18 P=6
19 GO TO (10,20,30,40,50,60,70,80,90),NCALL
20 CONTINUE
21 WRITE(P,100)
22 100 FORMAT(1H1,30X,'**TABLE 11. SUMMARY RESULTS**')
23 WRITE(P,102)(TITLE(I),I=1,20)
24 102 FORMAT(//10X,'CASE STUDY: ',20A4)
25 WRITE(P,103)
26 103 FORMAT(//10X,'INITIAL CONDITIONS:',T60,'800STER',T70,'ORBITER')
27 WRITE(P,104)NB00S,NORBI
28 104 FORMAT(15X,'NUMBER OF VEHICLE STAGES ',T63,I2,T73,I2)
29 MSAB=MGTHIU(JS2(1))+MGTH2U(JS1(1))
30 MSAB=MGTHIU(JS2(10))+MGTH2U(JS1(10))
31 WRITE(P,170)MSAB,MSAO
32 170 FORMAT(15X,'SERVICE CAPACITY OF SAFING AREA',T63,I2,T73,I2)
33 MFB=MGTHIU(JS2(2))+MGTH2U(JS1(2))
34 MFO=MGTHIU(JS2(11))+MGTH2U(JS1(11))
35 WRITE(P,105)MFB,MFO
36 105 FORMAT(15X,'SERVICE CAPACITY OF MAINTENANCE FACILITY' ,T63,I2,
37 1 T73,I2)
38 MCOTB=MGTHIU(JS2(3))+MGTH2U(JS1(3))
39 MCOTO=MGTHIU(JS2(12))+MGTH2U(JS1(12))
40 WRITE(P,176)MCOTB,MCOTO
41 176 FORMAT(15X,'SERVICE CAPACITY OF CHECKOUT & TEST FACILITY' ,T63,
42 1 I2,T73,I2)
43 MHF=MGTHIU(JS2(20))+MGTH2U(JS1(20))
44 WRITE(P,171)MHF
45 171 FORMAT(15X,'SERVICE CAPACITY OF MATE & HOOK-UP FACILITY',T68,I2)
46 LUTRF=MGTHIU(JS2(18))+MGTH2U(JS1(18))
47 WRITE(P,172)LUTRF
48 172 FORMAT(15X,'SERVICE CAPACITY OF LUT REFURBISHMENT FACILITY',T68,
49 1 I2)
50 WRITE(P,173)NLUTS
51 173 FORMAT(15X,'NUMBER OF LUTS-',T40,I2)
52 C*****
53 NPADS=1
54 IF(JF2(2).GT.0)NPADS=2
55 IF(JF2(3).GT.0)NPADS=3

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PRINT

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56 13 WRITE(P,106)NPAUS
57 106 FORMAT(15X,'NUMBER OF LAUNCH PADS=',I40,I2)
58 K2=JEKS(30)+JEKS(2)+JEKS(3)+JEKS(4)+JEKS(22)+JEKS(23)+JEKS(24)+
59 1 JEKS(25)+JEKS(8)
60 WRITE(P,174)
61 174 FORMAT(///10X,'TRAFFIC MODEL STATISTICS:1')
62 NFR=260000./FLOAT(JEKS(21))*FLOAT(K2)
63 WRITE(P,157)NFR
64 157 FORMAT(15X,'MISSION MODEL LAUNCH RATE (FLIGHTS/YEAR)=-',I6)
65 NLR=NELT1+NELT2+NELT3+NELT4+NELT5
66 PELT1=FLOAT(NELT1)*100/NLR
67 PELT2=FLOAT(NELT2)*100/NLR
68 PELT3=FLOAT(NELT3)*100/NLR
69 PELT4=FLOAT(NELT4)*100/NLR
70 PELT5=FLOAT(NELT5)*100/NLR
71 WRITE(P,158)PELT1,PELT2,PELT3,PELT4,PELT5
72 158 FORMAT(15X,'DISTRIBUTION OF TIME BETWEEN LAUNCH REQUESTS-',
73 16X,F5.1,'% OF LAUNCHES ARE SCHEDULED 1 DAY APART',
74 16X,F5.1,'% OF LAUNCHES ARE SCHEDULED 2 DAYS APART',
75 16X,F5.1,'% OF LAUNCHES ARE SCHEDULED 3 DAYS TO 1 WEEK APART',
76 16X,F5.1,'% OF LAUNCHES ARE SCHEDULED 1 WEEK TO 2 WEEKS APART',
77 16X,F5.1,'% OF LAUNCHES ARE SCHEDULED 2 WEEKS TO 1 MONTH APART')
78 WRITE(P,107)K2
79 107 FORMAT(///10X,'OPERATIONAL EVALUATION PARAMETERS FOR ',I4,' SIMULA
80 TIONS:')
81 PLR=FLOAT(JEKS(30))*100./K2
82 WRITE(P,152)JEKS(30),PLR
83 152 FORMAT(15X,'THE NUMBER OF SUCCESSFUL LAUNCH REQUESTS ARE',I4,
84 1 ' OR',F5.1,' PERCENT')
85 NMLR=K2-JEKS(30)
86 PNMLR=FLOAT(NMLR)*100./K2
87 WRITE(P,108)NMLR,PNMLR
88 108 FORMAT(15X,'THE NUMBER OF UNSUCCESSFUL REQUESTS ARE',I4,
89 1 ' OR',F5.1,' PERCENT')
90 GLOBP=(FLOAT(JEKS(26))/FLOAT(JEKS(30)))*100.
91 GLOB=JEKS(26)
92 WRITE(P,373)GLOB,GLOBP
93 373 FORMAT(15X,'THE NUMBER OF VEHICLES LAUNCHED AT A SUBSEQUENT WINDOW
94 1 IS',I3,' OR',F5.1,' PERCENT')
95 WRITE(P,596)
96 596 FORMAT(15X,'REASONS FOR REQUEST FAILURES-')
97 NBUNA=JEKS(13)
98 PBUNA=FLOAT(NBUNA)*100./K2
99 WRITE(P,109)NBUNA,PBUNA
100 109 FORMAT(17X,'BOOSTER UNAVAILABILITY',I50,I4,' TIMES OR',F5.1,' PERC
101 ENT')
102 NOUNA=JEKS(4)
103 POUNA=FLOAT(NOUNA)*100./K2
104 WRITE(P,110)NOUNA,POUNA
105 110 FORMAT(17X,'ORBITER UNAVAILABILITY',I50,I4,' TIMES OR',F5.1,' PERC
106 ENT')
107 PUNA=FLOAT(JEKS(21))*100./K2
108 WRITE(P,156)JEKS(21),PUNA
109 156 FORMAT(17X,'BOTH STAGES UNAVAILABLE',I50,I4,' TIMES OR',F5.1,' PE
110 RCENT')
111 LFAIL=JEKS(8)

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112 PLFAIL=FLCAT(LFAIL)*100./K2
113 WRITE(P,175)LFALL,PLFAIL
114 175 FORMAT(17X,'LUT UNAVAILABILITY',T50,14,' TIMES OR',F5,1,' PERCENT
115 1 ')
116 NPUAU=JEKS(25)
117 PPVAV=FLCAT(NPUAU)*100./K2
118 WRITE(P,150)NPUAU,PPVAV
119 150 FORMAT(17X,'PAD UNAVAILABILITY',T50,14,' TIMES OR',F5,1,' PERCENT
120 1 ')
121 NFAIL=JEKS(22)*JEKS(24)+JEKS(23)
122 PFAIL=FLOAT(NFAIL)*100./K2
123 WRITE(P,151)NFAIL,PFAIL
124 151 FORMAT(17X,'FAILURES AND REMOVAL FROM PAD',T50,14,' TIMES OR',
125 1 ,F5,1,' PERCENT')
126 WRITE(P,111)
127 111 FORMAT(///10X,'AVERAGE PHASE TIMES PER VEHICLE ELEMENT:',T60,
128 1 ,F8,0,'T70',F8,0,'ORBITER')
129 TLTSL=K4/1000.
130 TIME10=K5/1000.
131 RETURN
132 C*****
133 20 CONTINUE
134 TLTSL=K2/1000.
135 TLTSL=K3/1000.
136 WRITE(P,113)TLTSL,TLTSL
137 113 FORMAT(15X,'LANDING TO STORAGE(DAYS) ',T61,F5,2,T71,F5,2)
138 STORB=K4/1000.
139 STORB=K5/1000.
140 WRITE(P,114)STORB,STORB
141 114 FORMAT(15X,'STORAGE(DAYS) ',T61,F5,2,T71,F5,2)
142 WRITE(P,161)TLTSL,TLTSL
143 161 FORMAT(15X,'LEAVE STORAGE TO LAUNCH(DAYS)',T61,F5,2,T71,F5,2)
144 WRITE(P,160)TIME10
145 160 FORMAT(15X,'MEAN ON-ORBIT MISSION TIME(DAYS)',T71,F5,2)
146 TOTAL=TLTSL+STORB+TLTSL
147 TLTAL=TLTSL+STORB+TLTSL+TIME10
148 WRITE(P,112)TOTAL,TOTAL
149 112 FORMAT(15X,'TURNAROUND- LAUNCH TO LAUNCH(DAYS)',T61,F5,2,T71,F5,2)
150 RETURN
151 C*****
152 30 CONTINUE
153 WRITE(P,180)
154 180 FORMAT(1H)
155 WRITE(P,115)
156 115 FORMAT(///10X,'WAITING STATISTICS:
157 1 ,T60',F8,0,'BOOSTER',T70,'ORBITER')
158 SAFB = K2/1000.
159 SAFO = K3/1000.
160 PDES8 = FLOAT(JQ5(11)-JQ6(11))*100./JQ5(11)
161 PDES0 = FLOAT(JQ5(10)-JQ6(10))*100./JQ5(10)
162 WRITE(P,300)PDES8,PDES0,SAFB,SAFO
163 300 FORMAT(15X,'PERCENT OF VEHICLES DELAYED FOR SAFING',T62,F5,2,T72,F
164 15,2,'/15X,'AVERAGE QUEUE TIME FOR SAFING(DAYS)',T62,F5,2,T72,F5,2 )
165 PQMB=FLOAT(JQ5(12)-JQ6(12))*100./JQ5(12)
166 PQMO=FLOAT(JQ5(11)-JQ6(11))*100./JQ5(11)
167 WRITE(P,154)PQMB,PQMO

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168 154 FORMAT(15X,'PERCENT OF VEHICLES DELAYED FOR MAINT.',T62,F5.2,T72,
169 1 F5.2)
170 TMQB=K4/1000.
171 TMQ=K5/1000.
172 WRITE(P,116)TMQB,TMQ
173 116 FORMAT(15X,'AVERAGE QUEUE TIME FOR MAINTENANCE(DAYS)',T62,F5.2,T72
174 1 F5.2)
175 RETURN
176 C.....
177 40 CONTINUE
178 PCHEK8 = FLOAT(JQ5(3)-JQ6(3))*100./JQ5(3)
179 PCHEKO = FLOAT(JQ5(12)-JQ6(12))*100./JQ5(12)
180 CHECK8 = K2/1000.
181 CHECKO = K3/1000.
182 WRITE(P,330)PCHEK8,PCHEKO,CHECK8,CHEKO
183 330 FORMAT(
184 215X,'PERCENT OF VEHICLES DELAYED FOR CHECKOUT&TEST',T62,F5.2,T72,
185 3 F5.2,15X,'AVERAGE QUEUE TIME FOR CHECKOUT & TEST(DAYS)',T62,F5.2
186 42,T72,F5.2)
187 C WRITE(P,117) DUMMY WRITE STATEMENT
188 C 117 FORMAT(15X,'AVERAGE QUEUE TIME FOR
189 1 F5.2)
190 MAHUF = K4/1000.
191 LUTREF = K5/1000.
192 PMAHU = FLOAT(JQ5(20)-JQ6(20))*100./JQ5(20)
193 PLUT = FLOAT(JQ5(18)-JQ6(18))*100./JQ5(18)
194 WRITE(P,301)PMAHU,MAHUF,PLUT,LUTREF
195 301 FORMAT(15X,'PERCENT OF VEHICLES DELAYED FOR M & H',T65,F5.2,15X,
196 1 AVERAGE QUEUE TIME FOR MAINT & WOOKUP(DAYS)',T65,F5.2,15X,'PERCENT
197 2 OF LUTS DELAYED FOR REFURBISHMENT',T65,F5.2,15X,'AVERAGE QUEUE T
198 3IME FOR LUT REFURBISH FACILITY',T65,F5.2)
199 RETURN
200 C.....
201 50 CONTINUE
202 PQP=FLOAT(JQ5(21)-JQ6(21))*100./JQ5(21)
203 WRITE(P,155)PQP
204 155 FORMAT(15X,'PERCENT OF VEHICLES DELAYED FOR PAD ACCESS',T65,F5.2)
205 TQP=K2/1000.
206 WRITE(P,118)TQP
207 118 FORMAT(15X,'AVERAGE QUEUE TIME FOR PAD(DAYS)
208 RETURN
209 C.....
210 60 CONTINUE
211 SAUB=K2/10.
212 SAUB=K3/10.
213 WRITE(P,119)SAUB,SAUO
214 119 FORMAT(15X,'FACILITY UTILIZATION',T60,'BOOSTER',T70,'ORBITER',
215 1 /15X,'SAFING AREA(PERCENT)',T61,F5.2,T71,F5.2)
216 SMUB=K4/10.
217 SMUB=K5/10.
218 WRITE(P,120)SMUB,SMUO
219 120 FORMAT(15X,'MAINTENANCE(PERCENT)
220 RETURN
221 C.....
222 70 CONTINUE
223 COTUB=K2/10.

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224 COTUO=K3/10.
225 WRITE(P,177)COTUO,COTUO
226 177 FORMAT(15X,'CHECKOUT & TEST(PERCENT)',T61,F5.2,T71,F5.2)
227 RETURN
228 C.....
229 80 CONTINUE
230 PMHU=K2/10.
231 WRITE(P,121)PMHU
232 121 FORMAT(15X,'MATE & HOOK-UP(PERCENT)',T45,F5.2)
233 PLRU=K3/10.
234 WRITE(P,122)PLRU
235 122 FORMAT(15X,'LUI REFURBISHMENT(PERCENT)',T45,F5.2)
236 RETURN
237 C.....
238 90 CONTINUE
239 PDU1=K2/10.
240 WRITE(P,123)PDU1
241 123 FORMAT(10X,'PAD UTILIZATION',/
242 1 15X,'LAUNCH PAD 1 (PERCENT)',T45,F5.2)
243 IF (NPADS-2) 61,62,62
244 62 PDU2=K3/10.
245 WRITE(P,124)PDU2
246 124 FORMAT(15X,'LAUNCH PAD 2 (PERCENT)',T45,F5.2)
247 IF (NPADS-3) 61,1003,1003
248 1003 PDU3=K4/10.
249 WRITE(P,1004) PDU3
250 1004 FORMAT (15X,'LAUNCH PAD 3 (PERCENT) ',T45,F5.2)
251 61 CONTINUE
252 C.....WRITE OUT CASE STUDY COMMENT CARD
253 WRITE(P,199)(TITLE(1),JEL,20)
254 199 FORMAT(1H1,/'',20A4,'.....')
255 WRITE(P,200)NB005,NORB)
256 200 FORMAT(///,10X,'THE FLEET SIZE IS',I2,' BOOSTERS 6',I2,' ORBITERS
257 1')
258 WRITE(P,201)NSEED,NSEED)
259 201 FORMAT(10X,'RANDOM NUMBER SEED FOR TRAFFIC MODEL SIMULATOR = ',I10
260 1 ,/10X,'RANDOM NUMBER SEED FOR DISTRIBUTION FUNCTIONS = ',I10)
261 C
262 C DETERMINE IF TABLE III OUTPUT IS REQUIRED
263 IF (ISN.LT.9) RETURN
264 CALL SAMPLE(K1,K2,K3,K4,K5)
265 C
266 RETURN
267 C.....
268 END

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BHUG,P PROCS

REALN

420223*TPFS*REALN

1 SUBROUTINE REALN (RES,1IN)

2 C
3 C SUBROUTINE TO CHANGE ALPHANUMERIC NUMBERS TO REAL NUMBERS

4 C
5 C SUBSCRIPTED VARIABLE IA WILL CONTAIN 1 CHARACTER IN EACH POSITION

6 C
7 DIMENSION IA(6)

8 C
9 C

10 C
11 C FUNCTION FLD IS A LIBRARY FUNCTION WHICH ENABLES USER TO OBTAIN
12 C INDIVIDUAL BITS FROM AN ALPHANUMERIC WORD AND STORE
13 C THEM IN ANOTHER WORD.

14 C
15 C FLD (I,K,E) I = STARTING BIT K = BIT WIDTH E = WORD NAME

16 C
17 C

18 C
19 C IN POSITION 1 OF IA STORE BITS 0 TO 5 OF WORD IN
20 C IA(1) = FLD (0,6,IN)

21 C
22 C IN POSITION 2 OF IA STORE BITS 6 TO 11 OF WORD IN
23 C IA(2) = FLD (6,6,IN)

24 C
25 C IN POSITION 3 OF IA STORE BITS 12 TO 17 OF WORD IN
26 C IA(3) = FLD (12,6,IN)

27 C
28 C IN POSITION 4 OF IA STORE BITS 18 TO 23 OF WORD IN
29 C IA(4) = FLD (18,6,IN)

30 C
31 C IN POSITION 5 OF IA STORE BITS 24 TO 29 OF WORD IN
32 C IA(5) = FLD (24,6,IN)

33 C
34 C IN POSITION 6 OF IA STORE BITS 30 TO 35 OF WORD IN
35 C IA(6) = FLD (30,6,IN)

36 C
37 C TEST TO DETERMINE IF 1ST CHARACTER OF WORD IS A K
38 C IF (IA(1).NE.16) GO TO 1

39 C
40 C TEST TO DETERMINE IF 6TH CHARACTER OF WORD IS A BLANK
41 C IF (IA(6).NE. 5) GO TO 10

42 C
43 C TEST TO DETERMINE IF 5TH CHARACTER OF WORD IS A BLANK
44 C IF (IA(5).NE. 5) GO TO 11

45 C
46 C TEST TO DETERMINE IF 4TH CHARACTER OF WORD IS A BLANK
47 C IF (IA(4).NE. 5) GO TO 12

48 C
49 C TEST TO DETERMINE IF 3RD CHARACTER OF WORD IS A BLANK
50 C IF (IA(3).NE. 5) GO TO 13

51 C
52 C TEST TO DETERMINE IF 2ND CHARACTER OF WORD IS A BLANK
53 C IF (IA(2).NE. 5) GO TO 14

54 GO TO 1

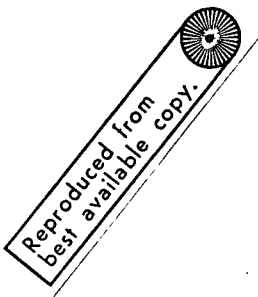
55 C

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REALN

56	C	
57	C	
58	C	TO CHANGE ALPHANUMERIC NUMBER TO A REAL NUMBER SUBTRACT 48.
59	C	
60	C	ALPHANUMERIC ZERO IS REPRESENTED BY CHARACTER 48 WHICH
61	C	OCCUPIES 6 BITS IN AN ALPHANUMERIC 40RD.
62	C	
63	C	ALPHANUMERIC ONE IS REPRESENTED BY CHARACTER 49 ETC.
64	C	
65	C	
66	C	
67	C	10 CONTINUE
68	C	
69	C	EQUATION TO CHANGE 5 CHARACTER ALPHANUMERIC NUMBER TO 5
70	C	CHARACTER INTEGER
71	C	$ires = (IA(6)-48) + (IA(5)-48)*10 + (IA(4)-48)*100$
72	C	$1 + (IA(3)-48)*1000 + (IA(2)-48)*10000$
73	C	
74	C	EQUATION TO CHANGE INTEGER TO REAL NUMBER FOR TABLE
75	C	$res = ires$
76	C	RETURN
77	C	11 CONTINUE
78	C	
79	C	EQUATION TO CHANGE 4 CHARACTER ALPHANUMERIC NUMBER TO 4
80	C	CHARACTER INTEGER
81	C	$ires = (IA(5)-48) + (IA(4)-48)*10 + (IA(3)-48)*100$
82	C	$1 + (IA(2)-48)*1000$
83	C	
84	C	EQUATION TO CHANGE INTEGER TO REAL NUMBER FOR TABLE
85	C	$res = ires$
86	C	RETURN
87	C	12 CONTINUE
88	C	
89	C	EQUATION TO CHANGE 3 CHARACTER ALPHANUMERIC NUMBER TO 3
90	C	CHARACTER INTEGER
91	C	$ires = (IA(4)-48) + (IA(3)-48)*10 + (IA(2)-48)*100$
92	C	
93	C	EQUATION TO CHANGE INTEGER TO REAL NUMBER FOR TABLE
94	C	$res = ires$
95	C	RETURN
96	C	13 CONTINUE
97	C	
98	C	EQUATION TO CHANGE 2 CHARACTER ALPHANUMERIC NUMBER TO 2
99	C	CHARACTER INTEGER
100	C	$ires = (IA(3)-48) + (IA(2)-48)*10$
101	C	
102	C	EQUATION TO CHANGE INTEGER TO REAL NUMBER FOR TABLE
103	C	$res = ires$
104	C	RETURN
105	C	14 CONTINUE
106	C	
107	C	EQUATION TO CHANGE 1 CHARACTER ALPHANUMERIC NUMBER TO 1
108	C	CHARACTER INTEGER
109	C	$ires = IA(2)-48$
110	C	
111	C	EQUATION TO CHANGE INTEGER TO REAL NUMBER FOR TABLE

B-36



112	RES = IRES
113	RETURN
114	1 CONTINUE
115	C
116	C EQUATION WILL RESULT IN ERROR, PRINTING STARS
117	RES = 9999999.9
118	RETURN
119	END

BHGG,P	RLOGN
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GPRT,S	RLOGN
FURPUR HA18-06/30-13:31	

RLOGN

DATE 063072

PAGE

1

420223*TPFS-RLOGN

1 FUNCTION RLOGN (J)

LGNS 10

2 C

3 C*****THE PARAMETERS USED WITH RLOGN ARE THE MEAN AND STANDARD DEVIATION

LGNS 20

4 C*****OF A NORMAL DISTRIBUTION

LGNS 30

5 C

6 VA= RNORM (J)

LGNS 40

7 RLOGN=EXP(VA)

LGNS 50

8 RETURN

9 END

@HUG,P RNORM

@PRT,S RNORM

FURPUR HA18-06/30-13:31

420223*YFFS.RNORM
1. FUNCTION RNORM (J)
2. COMMON /HEPL/PARAM(7,3),LL,ISEED,ISEED1,ISTORE(2,10),IS(2)
3. R8=DRAND1(ISEED1)
4. R8=DRAND1(ISEED1)
5. V=(-2.0*ALOG(R81))*0.5*COS(.6283185)
6. RNORM = V*PARAM (J,2) + PARAM (J,1)
7. RETURN
8. END

NRLS 10

NRLS 80
NRLS 90
NRLS 150
NRLS 160

0HDG,P SAMPLE
0PRT,S SAMPLE
FURPUR HA18-06/30-13:31

SAMPLE

420223*TPFS,SAMPLE		COMPILER(DATA=SHORT)		SAMP 10	
1		SUBROUTINE SAMPLE(K1,K2,K3,K4,K5)		SAMP 20	
2		COMMON /ISTAT/ NCONF		SAMP 30	
4		COMMON/STAT2/NFLTS		SAMP 40	
5		COMMON/STATS/NHEAD		SAMP 50	
6		COMMON/WAY/P		SAMP 60	
7		COMMON/STAT3/1SN		SAMP 70	
8		COMMON K(100)		SAMP 80	
9		EQUIVALENCE _IK(28),NCLOCK		SAMP 90	
10		INTEGER P		SAMP 100	
11		DIMENSION I(32,2),SAVE(14,6),NHEAD(14,11)		SAMP 110	
12	C			SAMP 120	
13	C			SAMP 130	
14	C	THESE ARE VALUES OF THE STUDENT T VARIABLE		SAMP 140	
15	C			SAMP 150	
16		DATA((T(J,K),K=1,2),J=1,18)/		SAMP 160	
17		112,206,63,657,		SAMP 170	
18		2 4,303, 9,925,		SAMP 180	
19		3 3,182, 5,841,		SAMP 190	
20		4 2,776, 4,604,		SAMP 200	
21		5 2,571, 4,032,		SAMP 210	
22		6 2,447, 3,707,		SAMP 220	
23		7 2,365, 3,499,		SAMP 230	
24		8 2,306, 3,358,		SAMP 240	
25		9 2,262, 3,250,		SAMP 250	
26		0 2,228, 3,169,		SAMP 260	
27		1 2,201, 3,106,		SAMP 270	
28		2 2,179, 3,055,		SAMP 280	
29		3 2,160, 3,012,		SAMP 290	
30		4 2,145, 2,977,		SAMP 300	
31		5 2,131, 2,947,		SAMP 310	
32		6 2,120, 2,921,		SAMP 320	
33		7 2,110, 2,898,		SAMP 330	
34		8 2,101, 2,878,		SAMP 340	
35		DATA((T(J,K),K=1,2),J=19,31)/		SAMP 350	
36		9 2,093, 2,861,		SAMP 360	
37		0 2,086, 2,845,		SAMP 370	
38		1 2,080, 2,831,		SAMP 380	
39		2 2,074, 2,819,		SAMP 390	
40		3 2,069, 2,807,		SAMP 400	
41		4 2,064, 2,797,		SAMP 410	
42		5 2,060, 2,787,		SAMP 420	
43		6 2,056, 2,779,		SAMP 430	
44		7 2,052, 2,771,		SAMP 440	
45		8 2,048, 2,763,		SAMP 450	
46		9 2,045, 2,756,		SAMP 460	
47		0 2,042, 2,750,		SAMP 470	
48		1 1,960, 2,576,		SAMP 480	
49	C			SAMP 490	
50	C			SAMP 500	
51	C	STORE HEADINGS TO BE USED IN PRINTING		SAMP 510	
52	C			SAMP 520	
53		DATA(INHEAD(I,J),J=1,11),I=1,9)/		SAMP 530	
54	C			SAMP 540	
55		162H THE PERCENTAGE OF SUCCESSFUL LAUNCH REQUESTS		SAMP 550	

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SAMPLE	DATE 063072	PAGE 2
56	262H THE PERCENTAGE OF UNSUCCESSFUL LAUNCH REQUESTS	SAMP 560
57	362H THE PERCENTAGE OF VEHICLES LAUNCHED AT A SUBSEQUENT WINDOW#	SAMP 570
58	462H BOOSTER UNAVAILABILITY	SAMP 580
59	562H ORBITER UNAVAILABILITY	SAMP 590
60	662H BOTH STAGES UNAVAILABLE	SAMP 600
61	762H LUT UNAVAILABILITY	SAMP 610
62	862H PAD UNAVAILABILITY	SAMP 620
63	962H FAILURES AND REMOVAL FROM PAD	SAMP 630
64	C	SAMP 640
65	C	SAMP 650
66	C	SAMP 660
67	C	SAMP 670
68	C	SAMP 680
69	C	SAMP 690
70	C	SAMP 700
71	C	SAMP 710
72	C	SAMP 720
73	C	SAMP 730
74	C	SAMP 740
75	C	SAMP 750
76	C	SAMP 760
77	C	SAMP 770
78	C	SAMP 780
79	C	SAMP 790
80	C	SAMP 800
81	C	SAMP 810
82	C	SAMP 820
83	C	SAMP 830
84	C	SAMP 840
85	C	SAMP 850
86	C	SAMP 860
87	C	SAMP 870
88	C	SAMP 880
89	C	SAMP 890
90	C	SAMP 900
91	C	SAMP 910
92	C	SAMP 920
93	C	SAMP 930
94	C	SAMP 940
95	C	SAMP 950
96	C	SAMP 960
97	C	SAMP 970
98	C	SAMP 980
99	C	SAMP 990
100	C	SAMP 1000
101	C	SAMP 1010
102	C	SAMP 1020
103	C	SAMP 1030
104	C	SAMP 1040
105	C	SAMP 1050
106	C	SAMP 1060
107	C	SAMP 1070
108	C	SAMP 1080
109	C	SAMP 1090
110	C	SAMP 1100
111	C	SAMP 1110

B-41

SAMPLE

112	C		SAMPLE1120
113	C	IDENTIFY THE LEVEL OF SIGNIFICANCE	SAMPLE1130
114	C	IF(NC1.EQ.99) NCONF=2	SAMPLE1140
115	C	IF(NC1.NE.99) NCONF=1	SAMPLE1150
116	C	CONTINUE	SAMPLE1160
117	C		SAMPLE1170
118	C	CHANGE FIXED POINT VARIABLES K4 AND K5 TO FLOATING POINT	SAMPLE1180
119	C	FK4=FLOAT(K4)	SAMPLE1190
120	C	FK5=FLOAT(K5)	SAMPLE1200
121	C		SAMPLE1210
122	C	SPECIFY THE K5 VALUE BASED ON THE TYPE OF STATISTIC BEING	SAMPLE1220
123	C	COLLECTED AS SPECIFIED BY K3	SAMPLE1230
124	C	IF(K3.EQ.1) FK5=1.	SAMPLE1240
125	C	IF(K3.EQ.2) FK5=FLOAT(CLOCK)/1000.	SAMPLE1250
126	C		SAMPLE1260
127	C	ADJUST THE K4 VALUE BASED ON THE TYPE OF STATISTIC BEING	SAMPLE1270
128	C	COLLECTED AS SPECIFIED BY K3	SAMPLE1280
129	C	IF(K3.GE.2) FK4=FK4*FK5/1000.	SAMPLE1290
130	C		SAMPLE1300
131	C	COUNT THE NUMBER OF OBSERVATIONS THAT HAVE BEEN MADE	SAMPLE1310
132	C	IF(K2.LE.1) NOR=NOR+1	SAMPLE1320
133	C		SAMPLE1330
134	C		SAMPLE1340
135	C	COLLECT AND RETAIN THE STATISTICS FROM THE FIRST 100 SIMS	SAMPLE1350
136	C	IF(NOB.LT.2) SAVE(K2,4)=FK4	SAMPLE1360
137	C	IF(NOB.LT.2) SAVE(K2,6)=FK5	SAMPLE1370
138	C	IF(NOB.EQ.1) RETURN	SAMPLE1380
139	C		SAMPLE1390
140	C		SAMPLE1400
141	C	DETERMINE THE CHANGE IN MEAN FROM THE CURRENT OBSERVATION AND THE	SAMPLE1410
142	C	100TH OBSERVATION	SAMPLE1420
143	C	XK2=FK4-SAVE(K2,4)	SAMPLE1430
144	C		SAMPLE1440
145	C	DETERMINE AND RETAIN THE DIFFERENCES IN MEANS BETWEEN THE CURRENT	SAMPLE1450
146	C	OBSERVATION AND THE PREVIOUS OBSERVATION TO OBTAIN A NEW MEAN	SAMPLE1460
147	C	X1=XK2=SAVE(K2,1)	SAMPLE1470
148	C	SAVE(K2,1)=XK2	SAMPLE1480
149	C	XK2=X1	SAMPLE1490
150	C		SAMPLE1500
151	C	DETERMINE THE INCREMENT IN BLOCK ENTRIES THAT HAVE OCCURRED	SAMPLE1510
152	C	SINCE THE PREVIOUS OBSERVATION	SAMPLE1520
153	C	BINCR=FK5-SAVE(K2,6)	SAMPLE1530
154	C		SAMPLE1540
155	C	RETAIN THE TOTAL NUMBER OF BLOCK ENTRIES	SAMPLE1550
156	C	SAVE(K2,6)=FK5	SAMPLE1560
157	C		SAMPLE1570
158	C		SAMPLE1580
159	C	ADJUST THE NEW MEAN BASED ON THE TYPE OF STATISTIC BEING	SAMPLE1590
160	C	COLLECTED AS SPECIFIED BY K3	SAMPLE1600
161	C	IF(K3.EQ.1) XK2=XK2*2.	SAMPLE1610
162	C	IF(K3.EQ.2) XK2=XK2*100/BINCR	SAMPLE1620
163	C	IF(K3.EQ.3) XK2=XK2/11NCR	SAMPLE1630
164	C		SAMPLE1640
165	C		SAMPLE1650
166	C		SAMPLE1660
167	C	DETERMINE THE 1 STATISTIC VALUE AND THE ESTIMATED	SAMPLE1670

SAMPLE

169	C	POPULATION STANDARD DEVIATION	SAMP1680
169		SAVE(K2,2) = SAVE(K2,2) + XK2	SAMP1690
170		SAVE(K2,3) = SAVE(K2,3) + (XK2*XK2)	SAMP1700
171	C		SAMP1710
172	C	IF AN INSUFFICIENT NUMBER OF OBSERVATIONS HAS OCCURRED, RETURN	SAMP1720
173		N = NOB	SAMP1730
174		(FIN.LE.2) RETURN	SAMP1740
175	C		SAMP1750
176	C	DETERMINE THE DEGREES OF FREEDOM FOR THE STUDENT T DISTRIBUTION	SAMP1760
177		NDFREE = N-2	SAMP1770
178		IF(NDFREE.GT.31) NDFREE = 31	SAMP1780
179	C		SAMP1781
180	C	DETERMINE AND RETAIN THE ESTIMATED DEVIATION	SAMP1782
181		CALL STDV(SAVE(K2,2),SAVE(K2,3),N,STDEV)	SAMP1790
182		SAVE(K2,5)=T(INDFREE,ACONF)*STDEV	SAMP1800
183	C		SAMP1810
184		RETURN	SAMP1820
185	C		SAMP1830
186	C		SAMP1840
187	C		SAMP1850
188	C		SAMP1860
189	C		SAMP1870
190	C		SAMP1880
191	C		SAMP1890
192	60	CONTINUE	SAMP1900
193	C		SAMP1910
194	C	BEGIN OUTPUT OF CONFIDENCE INTERVALS	SAMP1920
195	C		SAMP1930
196	C		SAMP1940
197	C		SAMP1950
198	C		SAMP1960
199	C	OUTPUT FOR TABLE III HEADINGS	SAMP1970
200	C		SAMP1980
201		WRITE(P,100)	SAMP1990
202	100	FORMAT(IH1,40X,***TABLE III. STATISTICAL RESULTS***)	SAMP2000
203		WRITE(P,110)	SAMP2010
204	110	FORMAT(///35X,CONFIDENCE INTERVALS ON REQUESTED STATISTICS*)	SAMP2020
205		WRITE(P,120)	SAMP2030
206	120	FORMAT(//25X,*WARNING = CONFIDENCE INTERVALS WITH LIMITS OF 0 OR 10	SAMP2040
207		10. ARE PROBABLY*)	SAMP2050
208		WRITE(P,130) NCI	SAMP2060
209	130	FORMAT(35X,*NOT,13,* PERCENT CONFIDENCE INTERVALS*)	SAMP2070
210		WRITE(P,140) N	SAMP2080
211	140	FORMAT(//35X,*STATISTICS BASED ON,13,* OBSERVATIONS,*/))	SAMP2090
212		WRITE(P,160) NCI	SAMP2100
213	160	FORMAT(//83,12,* PERCENT CONFIDENCE INTERVALS*	SAMP2110
214		1//79,*LOWER LIMIT,7X,*MEAN,6X,*UPPER LIMIT*)	SAMP2120
215		WRITE(P,150) NFLTS	SAMP2130
216	150	FORMAT(///5X,*STATISTICAL CONFIDENCE INTERVALS FOR *,14,* SIMULATI	SAMP2140
217		IONS.)*	SAMP2150
218	C		SAMP2160
219	C		SAMP2170
220	C	OUTPUT FOR THE OPERATIONAL EVALUATION PARAMETERS	SAMP2180
221	C		SAMP2190
222		DO 50 I=1,9	SAMP2200
223		AVG=SAVE(I,21)/FLOAT(N)	SAMP2210

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224	XLOW=AVG=SAVE(1,5)	SAMP2220
225	HIGH=AVG+SAVE(1,5)	SAMP2230
226	IF(HIGH,GE,100.) HIGH=100.	SAMP2240
227	IF(XLOW,LE,0.) XLOW=0.0	SAMP2250
228	WRITE(P,200)(NHEAD(1,J),J=1,11),XLOW,AVG,HIGH	SAMP2260
229	200 FORMAT(10X,11A6,1B1,F6.2,8X,F6.2)	SAMP2270
230	C	SAMP2280
231	IF(1,NE,3) GO TO 50	SAMP2290
232	WRITE(P,190)	SAMP2300
233	190 FORMAT(12X,'REASONS FOR REQUEST FAILURES:')	SAMP2310
234	C	SAMP2320
235	50 CONTINUE	SAMP2330
236	C	SAMP2340
237	C	SAMP2350
238	C	SAMP2360
239	DETERMINE IF ANY USER REQUESTED STATISTICS HAVE BEEN REQUESTED	SAMP2370
240	IF(1,IS,9) GO TO 500	SAMP2380
241	C	SAMP2390
242	C	SAMP2400
243	C	SAMP2410
244	WRITE(P,210)	SAMP2420
245	210 FORMAT(12X,'THE FOLLOWING ARE THE USER REQUESTED STATISTICS:')	SAMP2430
246	DO 52 I=10,15N	SAMP2440
247	AVG=SAVE(1,21)/ELQAT(N)	SAMP2450
248	XLOW=AVG=SAVE(1,5)	SAMP2460
249	HIGH=AVG+SAVE(1,5)	SAMP2470
250	IF(HIGH,GE,100.) HIGH=100.0	SAMP2480
251	IF(XLOW,LE,0.) XLOW=0.0	SAMP2490
252	WRITE(P,200)(NHEAD(1,J),J=1,11),XLOW,AVG,HIGH	SAMP2500
253	52 CONTINUE	SAMP2510
254	C	SAMP2520
255	500 WRITE(P,999)	SAMP2530
256	999 FORMAT(1H1)	SAMP2540
257	C	SAMP2550
258	RETURN	SAMP2560
259	END	SAMP2570

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BMDG,P	STAT
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BPRT,S	STAT
FURPUR	HA18-06/30-13:31

420223*TPFS*STAT			STAT 10
1	SUBROUTINE STAT		STAT 20
2	COMMON/STATS/NHEAD		STAT 30
3	DIMENSION NHEAD(14,11),ICOM(11)		STAT 40
4	COMMON/STAT3/ISN		STAT 50
5	DATA KEY/4HCONE/,JEND/5HIART /		
6	C		STAT 60
7	C		STAT 70
8	C		STAT 80
9	1	BEGIN SEARCH FOR A STATISTICAL REQUEST	STAT 90
10		CONTINUE	STAT 100
11	100	READ(5,100) ICOM,IRES,NUM	STAT 110
12	C	FORMAT(1X,10A6,A5,A4,A1)	STAT 120
13	C		STAT 130
14	C	IS THIS THE LAST CARD	STAT 140
15	C	IF THIS IS THE LAST CARD, RETURN	STAT 150
16	C	IF(ICOM(2).EQ.JEND) RETURN	STAT 160
17	C		STAT 170
18	C		STAT 180
19	C	NO,THIS IS NOT THE LAST CARD	STAT 190
20	C	IS THIS A STATISTICAL REQUEST	STAT 200
21	C	IF NOT, READ ANOTHER CARD	STAT 210
22	C	IF(KEY.NE.IRES) GO TO 1	STAT 220
23	C		STAT 230
24	C		STAT 240
25	C	THIS CARD IS A STATISTICAL REQUEST	STAT 250
26	C	CONVERT THE ALPHANUMERIC INPUT (NUM) TO NUMERIC	STAT 260
27	C	NUMISO	STAT 270
28	C	NUMISO=FLD(10,6,NUM)	STAT 280
29	C		STAT 290
30	C		STAT 300
31	C	DETERMINE THE STATISTICAL REQUEST IDENTIFICATION NUMBER	STAT 310
32	C	NUM=NUM1-48	STAT 320
33	C		STAT 330
34	C		STAT 340
35	C		STAT 350
36	C	DETERMINE IF THE NUMERIC IDENTIFICATION IS LEGAL	STAT 360
37	C	IF NOT, GO TO ERROR IDENTIFICATION ROUTINE	STAT 370
38	C	IF(NUM.LT.1.OR.NUM.GT.5) GO TO 3	STAT 380
39	C		STAT 390
40	C	STORE HEADING FOR FUTURE REFERENCE	STAT 400
41	C	ISN=NUM+9	STAT 410
42	C	DO 2 I=1,11	STAT 420
43	2	NHEAD(ISN,I)=ICOM(I)	STAT 430
44	C		STAT 440
45	C		STAT 450
46	C	READ ANOTHER CARD	STAT 460
47	C	GO TO 1	STAT 470
48	C		STAT 480
49	C		STAT 490
50	C	-----	STAT 500
51	C		STAT 510
52	C		STAT 520
53	C	ERROR IDENTIFICATION ROUTINE	STAT 530
54	C		STAT 540
55	3	J=J+1	STAT 550

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STAT

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56 IF(J,GT,1) GO TO 4          STAT 560
57 WRITE(6,300)                STAT 570
58 FORMAT(1H1)                  STAT 580
59 WRITE(6,310)                  STAT 590
60 FORMAT(10X,'THE FOLLOWING HEADINGS(S) ARE DISREGARDED IN THE CONFIDSTAT 600
61 FENCE INTERVAL/5X,'CALCULATIONS DUE TO AN ILLEGAL C. I. IDENTIFICATION 610
62 2TION NUMBER')              STAT 620
63 4 WRITE(6,320) (ICOM(L),1=1,11) STAT 630
64 320 FORMAT(15X,11A6)         STAT 640
65 C                             STAT 650
66 C                             STAT 660
67 READ ANOTHER CARD           STAT 670
68 GO TO 1                      STAT 680
69 END.                         STAT 690

```

BDUG,P STDV

BPRT,S STDV
FURPUR HA18-06/30-13:31

TABLE I

420223*YFES*TABLE I

```

1 SUBROUTINE TABLE I
2 C
3 C TABLE I IS A SUBROUTINE DESIGNED TO BE USED IN CONJUNCTION WITH
4 C A GPSS PROGRAM CARD DECK. THE FUNCTION OF THIS SUBROUTINE IS TO
5 C READ EACH CARD IN A GPSS DECK AND DETERMINE IF IT IS A COMMENT
6 C CARD WITH THE KEY WORD IN COLUMNS 68-71. (THE KEY WORD IS IDST)
7 C AFTER IT HAS FOUND A COMMENT CARD WITH THE PROPER KEY WORD IT THEN
8 C READS CARDS UNTIL IT FINDS A HELP CARD. AFTER THE HELP CARD IS FOUND
9 C THE SUBROUTINE OUTPUT IS CALLED TO PRINT OUT THE INFORMATION.
10 C IN TABULAR FORM
11 C
12 C DIMENSION NAME(2), ICOM(11)
13 C DATA JAST/IH*2,JHELP/4HHELP,KEY/6H IDST /,IEND/6H END /
14 C DATA JEND/6HSTART /
15 C
16 C WRITE(6,80)
17 C WRITE(6,81)
18 C WRITE(6,88)
19 C WRITE(6,70)
20 C WRITE(6,84)
21 C WRITE(6,82)
22 C WRITE(6,83)
23 C WRITE(6,84)
24 C WRITE(6,70)
25 C WRITE(6,84)
26 C WRITE(6,85)
27 C WRITE(6,85)
28 C
29 C BEGIN THE SEARCH FOR THE NEXT COMMENT CARD AT THIS POINT
30 C 5 CONTINUE
31 C READ(5,100) IAST,ICOM,IREM
32 C 100 FORMAT (A1,A5,2A6,9A6)
33 C
34 C IS THIS THE LAST CARD
35 C IF(ICOM(2),EQ,JEND) GO TO 999
36 C
37 C NO, THIS IS NOT THE LAST CARD
38 C IS THIS CARD A COMMENT CARD
39 C IF (IAST .NE. JAST) GO TO 5
40 C
41 C YES, THIS CARD IS A COMMENT CARD
42 C DOES THIS COMMENT CARD HAVE A KEY
43 C IF (IREM .NE. KEY) GO TO 5
44 C
45 C YES THIS COMMENT CARD HAS A KEY
46 C 10 CONTINUE
47 C READ(5,100) IAST,LOC,NAME,IX,IY,IZ,ISELM,NBA,NBB,MEAN,MOD,IREM
48 C WAS THE CARD JUST READ A HELP CARD
49 C IF (JHELP .NE. NAME(1)) GO TO 10
50 C
51 C YES, THE CARD JUST READ WAS A HELP CARD
52 C CALL OUTPUT AND PRINT APPROPRIATE INFORMATION
53 C CALL OUTPUT (ICOM,LOC,NAME,IX,IY,IZ,ISELM,NBA,NBB,MEAN,MOD)
54 C GO TO 5
55 C YES, THIS IS THE LAST CARD

```

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TABLE I		DATE 063072	PAGE 2
56	999 CONTINUE		
57	WRITE (6,70)		
58	WRITE (6,99)		
59	C		
60	70 FORMAT (1X,130H-----)		
61	1-----)		
62	2-----)		
63	C		
64	80 FORMAT (1H)		
65	C		
66	81 FORMAT (44X,*****TABLE I. OPERATION TIME DATA IN DAYS****)		
67	C		
68	82 FORMAT (1X,1H,29X,9HOPERATION,29X,1H,2X,		
69	1 9HDISIRIBU=,2X,1H,16X,14HINPUT DATA FOR,16X,1H)		
70	C		
71	83 FORMAT (1X,1H,28X,11HDESCRIPTION,28X,1H,2X,9HTION TYPE,2X,		
72	1 1H,12X,23HDISTRIBUTION PARAMETERS,11X,1H)		
73	C		
74	84 FORMAT (1X,1H,67X,1H,13X,1H,46X,1H)		
75	C		
76	85 FORMAT (1X,1H,67X,1H,13X,1H,14X,1H,17X,1H,13X,1H)		
77	C		
78	88 FORMAT (//)		
79	C		
80	99 FORMAT (1H)		
81	RETURN		
82	END		
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0H0G,P TRIAG			
0PRT,S TRIAG			
FURPUR HA18-06/30-13:31			

TRIAG

4202230YPF5,TRIAG

```

1 FUNCTION TRIAG(JP)
2 COMMON /HEPL/PARAM(7,3),LL,ISEED,ISEED1,ISTORE(2,10),IS(2)
3 RNUM = DRAND(1,ISEED)
4 IF (RNUM - (PARAM(JP,2) - PARAM(JP,1))) / (PARAM(JP,3) - PARAM(JP,1))) < 1,2
5 1 TRIAG = PARAM(JP,1) + SQRT((PARAM(JP,2) - PARAM(JP,1)))
6 1 * (PARAM(JP,2) - PARAM(JP,1)) * RNUM
7 RETURN
8 2 X = (PARAM(JP,3) - PARAM(JP,2)) * 2 - (PARAM(JP,2) - PARAM(JP,3)) *
9 1 (PARAM(JP,2) - PARAM(JP,1)) - (PARAM(JP,3) - PARAM(JP,1)) * RNUM
10 TRIAG = PARAM(JP,3) - SQRT(X)
11 RETURN
12 END

```

@HDG,P TRIAG

@PRT,S TRIAG
FURPUR HA18-06/30-13:31

TRIAGI

420223*TPFS*TRIAGI

```

1 FUNCTION TRIAGI(JP)
2 COMMON /HEPL/PARAM(7,3),LL,ISEED,ISEED1,ISTORE(2,10),IS(2)
3 RNUM = DRAND1(ISEED1)
4 IF(RNUM-(PARAM(JP,2)-PARAM(JP,1)))/(PARAM(JP,3)-PARAM(JP,1)))1,1,2
5 1 TRIAGI = PARAM(JP,1) + SQRT((PARAM(JP,3)-PARAM(JP,1))
6 1*(PARAM(JP,2)-PARAM(JP,1))*RNUM)
7 RETURN
8 2 X = (PARAM(JP,3)-PARAM(JP,2))*2 -(PARAM(JP,2)-PARAM(JP,3))*
9 1*(PARAM(JP,2)-PARAM(JP,1))-(PARAM(JP,3)-PARAM(JP,1))*RNUM
10 TRIAGI = PARAM(JP,3) - SQRT(X)
11 RETURN
12 END

```

QHDG,P UNFRM

GPRT,S UNFRM
FURPUR HA18-06/30-13:31

420223*TPFS,UNFRM

1	FUNCTION UNFRM (A,B)	UFMS 10
2	COMMON /HEPL/PARAM(7,3),LL,ISEED,ISEED1,ISTORE(2,10),IS(2)	
3	RNUM=DRAND(1,ISEED)	UFMS 60
4	UNFRM = A+(B-A)*RNUM	UFMS 70
5	RETURN	UFMS 80
6	END	UFMS 90

GDG,P UNFRM1

GPRT,S UNFRM1
FURPUR HA18-06/30-13:31

420223*PPFS*UNFRM1

1	FUNCTION UNFRM1(A,B)	UNFM 10
2	COMMON /HEPL/PARAM(7,3),LL,ISEED,ISEED1,ISTORE(2,10),IS(2)	UNFM 20
3	RNUM=DRAND1(ISEED1)	UNFM 30
4	UNFRM1 = A+(B-A)*RNUM	UNFM 40
5	RETURN	UNFM 50
6	END	UNFM 60

APPENDIX C. EXAMPLE TABLE1 OUTPUT

TABLE I. OPERATION TIME DATA IN DAYS

OPERATION DESCRIPTION	DISTRIBUTION TYPE	INPUT DATA FOR DISTRIBUTION PARAMETERS
MATE BOOSTER, ORBITER AND LUT	TRIANGULAR	MIN = 1.0000 MODE = 2.0000 MAX = 2.0000
COUNTDOWN PREPARATION	TRIANGULAR	MIN = 1.5000 MODE = 2.0000 MAX = 2.5000
FINAL COUNTDOWN	CONSTANT	MEAN = .0830
REMOVE POST LAUNCH LUT FROM PAD	TRIANGULAR	MIN = 1.2500 MODE = 1.5000 MAX = 1.7500
TRANSPORT LUT TO MAINT., REFURBISH/TEST, TRANSPORT TO STORAGE	TRIANGULAR	MIN = 2.7000 MODE = 3.0000 MAX = 3.3000
REFURBISH LAUNCH PAD	TRIANGULAR	MIN = .9500 MODE = 1.0000 MAX = 1.0500
LIFTOFF THROUGH SEPARATION	CONSTANT	MEAN = .0020
BOOSTER COAST, REENTRY, PLYBACK AND LANDING	TRIANGULAR	MIN = .0640 MODE = .0700 MAX = .0760
BOOSTER SAFING	TRIANGULAR	MIN = .9000 MODE = 1.0000 MAX = 1.1000
TRANSPORT BOOSTER TO MAINTENANCE	TRIANGULAR	MIN = .0500 MODE = .1000 MAX = .1500
BOOSTER MAINTENANCE	TRIANGULAR	MIN = 2.2500 MODE = 2.5000 MAX = 2.7500
BOOSTER ADDITIONAL MAINTENANCE	EXPONENTIAL	MIN = .5000 MEAN = 1.0000
TRANSPORT BOOSTER TO SYSTEMS TEST, TEST, TRANSPORT TO STORAGE	TRIANGULAR	MIN = 1.9000 MODE = 2.0000 MAX = 2.1000
ORBITER SAFING	TRIANGULAR	MIN = .9000 MODE = 1.0000 MAX = 1.1000
TRANSPORT ORBITER TO MAINTENANCE	TRIANGULAR	MIN = .0500 MODE = .1000 MAX = .1500
ORBITER MAINTENANCE	TRIANGULAR	MIN = 2.2500 MODE = 2.5000 MAX = 2.7500
ORBITER ADDITIONAL MAINTENANCE	EXPONENTIAL	MIN = .5000 MEAN = 1.0000
TRANSPORT ORBITER TO SYSTEMS TEST, TEST, TRANSPORT TO STORAGE	TRIANGULAR	MIN = 1.9000 MODE = 2.0000 MAX = 2.1000
REMOVE BOOSTER, ORBITER AND LUT FROM THE PAD	TRIANGULAR	MIN = 1.8000 MODE = 2.0000 MAX = 2.2000
TRANSPORT VEHICLE, SEPARATE ORBITER, TRANSPORT ORBITER TO MAINT	TRIANGULAR	MIN = .5500 MODE = .7500 MAX = .9500
SEPARATE BOOSTER FROM LUT, TRANSPORT BOOSTER TO MAINTENANCE	TRIANGULAR	MIN = .5500 MODE = .7500 MAX = .9500
BOOSTER MAINTENANCE ON RETURN FROM PAD	EXPONENTIAL	MIN = .5000 MEAN = 7.5000
BOOSTER VEHICLE SYSTEM TEST FACILITY ON RETURN FROM PAD	TRIANGULAR	MIN = 1.9000 MODE = 2.0000 MAX = 2.1000
ORBITER MAINTENANCE ON RETURN FROM PAD	EXPONENTIAL	MIN = .5000 MEAN = 7.5000
ORBITER VEHICLE SYS TEST FACILITY ON RETURN FROM PAD	TRIANGULAR	MIN = 1.9000 MODE = 2.0000 MAX = 2.1000

APPENDIX D. SAMPLE SOS MODEL OUTPUT

TABLE II. SUMMARY RESULTS

CASE STUDY: DETERMINATION OF FLEET SIZE REQUIREMENTS

INITIAL CONDITIONS:	BOOSTER	ORBITER
NUMBER OF VEHICLE STAGES	4	5
SERVICE CAPACITY OF SAFING AREA	1	1
SERVICE CAPACITY OF MAINTENANCE FACILITY	1	2
SERVICE CAPACITY OF CHECKOUT & TEST FACILITY	2	2
SERVICE CAPACITY OF MATE & HOOK-UP FACILITY	1	1
SERVICE CAPACITY OF LUT REFURBISHMENT FACILITY	2	2
NUMBER OF LUTS	3	
NUMBER OF LAUNCH PADS	2	

TRAFFIC MODEL STATISTICS:

MISSION MODEL LAUNCH RATE (FLIGHTS/YEAR) = 53
 DISTRIBUTION OF TIME BETWEEN LAUNCH REQUESTS =
 .8% OF LAUNCHES ARE SCHEDULED 1 DAY APART
 14.1% OF LAUNCHES ARE SCHEDULED 2 DAYS APART
 48.6% OF LAUNCHES ARE SCHEDULED 3 DAYS TO 1 WEEK APART
 35.3% OF LAUNCHES ARE SCHEDULED 1 WEEK TO 2 WEEKS APART
 1.2% OF LAUNCHES ARE SCHEDULED 2 WEEKS TO 1 MONTH APART

OPERATIONAL EVALUATION PARAMETERS FOR 302 SIMULATIONS:

THE NUMBER OF SUCCESSFUL LAUNCH REQUESTS ARE 484 OR 96.4 PERCENT
 THE NUMBER OF UNSUCCESSFUL REQUESTS ARE 18 OR 3.6 PERCENT
 THE NUMBER OF VEHICLES LAUNCHED AT A SUBSEQUENT WINDOW IS 7 OR 1.4 PERCENT
 REASONS FOR REQUEST FAILURES =
 BOOSTER UNAVAILABILITY 5 TIMES OR 1.0 PERCENT
 ORBITER UNAVAILABILITY 0 TIMES OR .0 PERCENT
 BOTH STAGES UNAVAILABLE 0 TIMES OR .0 PERCENT
 LUT UNAVAILABILITY 1 TIMES OR .2 PERCENT
 PAD UNAVAILABILITY 0 TIMES OR .0 PERCENT
 FAILURES AND REMOVAL FROM PAD 12 TIMES OR 2.4 PERCENT

AVERAGE PHASE TIMES PER VEHICLE ELEMENT:	BOOSTER	ORBITER
LANDING TO STORAGE (DAYS)	6.12	5.87
STORAGE (DAYS)	7.93	9.67
LEAVE STORAGE TO LAUNCH (DAYS)	5.32	5.32
MEAN ON-ORBIT MISSION TIME (DAYS)		3.42
TURNAROUND - LAUNCH TO LAUNCH (DAYS)	19.37	24.29

WAITING STATISTICS		BOOSTER	ORBITER
PERCENT OF VEHICLES DELAYED FOR SAFING		.00	3.93
AVERAGE QUEUE TIME FOR SAFING(DAYS)		.00	.39
PERCENT OF VEHICLES DELAYED FOR MAINT.		9.48	.61
AVERAGE QUEUE TIME FOR MAINTENANCE(DAYS)		2.69	.16
PERCENT OF VEHICLES DELAYED FOR CHECKOUT&TEST		.00	.00
AVERAGE QUEUE TIME FOR CHECKOUT & TEST(DAYS)		.00	.00
PERCENT OF VEHICLES DELAYED FOR M & H		.40	
AVERAGE QUEUE TIME FOR MATE & HOOKUP(DAYS)		.63	
PERCENT OF LUTS DELAYED FOR REFURBISHMENT		.20	
AVERAGE QUEUE TIME FOR LUT REFURBISH FACILITY		.00	
PERCENT OF VEHICLES DELAYED FOR PAD ACCESS		.00	
AVERAGE QUEUE TIME FOR PAD(DAYS)		.00	

FACILITY UTILIZATION		BOOSTER	ORBITER
SAFING AREA(PERCENT)		19.80	19.90
MAINTENANCE(PERCENT)		57.70	29.90
CHECKOUT & TEST(PERCENT)		20.30	20.30
MATE & HOOK-UP(PERCENT)		40.90	
LUT REFURBISHMENT(PERCENT)		30.50	
PAD UTILIZATION:			
LAUNCH PAD 1 (PERCENT)		68.70	
LAUNCH PAD 2 (PERCENT)		30.00	

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[illegible]

211	0	12	212	0	12	213	0	12	214	0	12	215	0	12
216	0	12	217	0	12	218	0	12	219	0	12	220	0	12
221	0	12	222	0	12	223	0	12	224	0	12	225	0	12
226	0	12	227	0	12	228	0	12	229	0	12	230	0	12
231	0	12	232	0	12	233	0	12	234	0	12	235	0	12
236	0	12	237	0	12	238	0	12	239	0	12	240	0	12
241	0	12	242	0	12	243	0	12	244	0	12	245	0	12
246	0	12	247	0	12	248	0	12	249	0	12	250	0	6
251	0	5	252	0	5	253	0	5	254	0	5	255	0	1
256	0	0	257	0	0	258	0	0	259	0	0	260	0	501
261	0	501	262	0	501	263	0	501	264	0	501	265	0	501
266	0	1	267	0	1	268	0	1	269	0	1	270	0	1
271	0	1	272	0	1	273	0	1	274	0	1	275	0	1
276	0	501	277	0	501	278	0	0	279	0	0	280	0	0

SAVE	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR
1	497	2	0	3	5	4	0	5	0	5	6	0	5	6
6	503	7	1	8	1	9	0	10	0	10	2549	0	10	2549
11	3280	12	1	13	4577	14	0	15	0	15	347	0	15	347
16	3280	17	1	18	4577	19	0	20	0	20	347	0	20	347
21	2931322	22	12	23	0	24	0	25	0	25	0	0	25	0
26	7	27	1	28	477	29	8	30	0	30	484	0	30	484

FACILITY	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR
1	497	2	0	3	5	4	0	5	0	5	6	0	5	6
2	503	7	1	8	1	9	0	10	0	10	2549	0	10	2549
11	3280	12	1	13	4577	14	0	15	0	15	347	0	15	347
16	3280	17	1	18	4577	19	0	20	0	20	347	0	20	347
21	2931322	22	12	23	0	24	0	25	0	25	0	0	25	0
26	7	27	1	28	477	29	8	30	0	30	484	0	30	484

STORAGE	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR
1	497	2	0	3	5	4	0	5	0	5	6	0	5	6
2	503	7	1	8	1	9	0	10	0	10	2549	0	10	2549
11	3280	12	1	13	4577	14	0	15	0	15	347	0	15	347
16	3280	17	1	18	4577	19	0	20	0	20	347	0	20	347
21	2931322	22	12	23	0	24	0	25	0	25	0	0	25	0
26	7	27	1	28	477	29	8	30	0	30	484	0	30	484

QUEUE	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR	NR
1	497	2	0	3	5	4	0	5	0	5	6	0	5	6
2	503	7	1	8	1	9	0	10	0	10	2549	0	10	2549
11	3280	12	1	13	4577	14	0	15	0	15	347	0	15	347
16	3280	17	1	18	4577	19	0	20	0	20	347	0	20	347
21	2931322	22	12	23	0	24	0	25	0	25	0	0	25	0
26	7	27	1	28	477	29	8	30	0	30	484	0	30	484

2	3	.05	496	499	90.52	254.53	2686.09	0	0
3	1	.00	495	495	100.00	.00	.00	0	0
10	1	.00	483	464	96.07	15.54	395.00	0	0
11	1	.00	495	492	99.39	.96	159.00	0	0
12	1	.00	494	494	100.00	.00	.00	0	0
18	1	.00	495	494	99.80	.00	.00	0	0
20	1	.00	497	495	99.60	2.54	631.50	0	0
21	1	.00	496	496	100.00	.00	.00	0	0
29	1	.00	503	503	100.00	.00	.00	0	0
31	1	.00	503	503	100.00	.00	.00	0	0

TABLE NUMBER 1

ENTRIES IN TABLE
483STANDARD DEVIATION
1634.808

NON-WEIGHTED

UPPER LIMIT	OBSERVED FREQUENCY	PERCENT OF TOTAL	CUMULATIVE PERCENTAGE	CUMULATIVE REMAINDER	MULTIPLE OF MEAN	DEVIATION FROM MEAN
5000	0	.00	.0	100.0	.817	-.686
5500	74	15.32	15.3	84.7	.899	-.380
6000	264	54.66	70.0	30.0	.980	-.074
6500	63	13.04	83.0	17.0	1.062	.232
7000	39	8.07	91.1	8.9	1.144	.538
7500	18	3.73	94.8	5.2	1.225	.843
8000	10	2.07	96.9	3.1	1.307	1.149
8500	0	.00	96.9	3.1	1.389	1.455
9000	4	.83	97.7	2.3	1.470	1.761
9500	1	.21	97.9	2.1	1.552	2.067
10000	0	.00	97.9	2.1	1.634	2.373
10500	1	.21	98.1	1.9	1.715	2.678
11000	2	.41	98.6	1.4	1.797	2.984
11500	0	.00	98.6	1.4	1.879	3.290
12000	1	.21	98.8	1.2	1.960	3.596
12500	0	.00	98.8	1.2	2.042	3.902
13000	0	.00	98.8	1.2	2.124	4.208
13500	0	.00	98.8	1.2	2.205	4.514
14000	0	.00	98.8	1.2	2.287	4.819
OVERFLOW	6	1.24	100.0	.0		

TABLE NUMBER 2

ENTRIES IN TABLE
497STANDARD DEVIATION
3142.907

NON-WEIGHTED

UPPER LIMIT	OBSERVED FREQUENCY	PERCENT OF TOTAL	CUMULATIVE PERCENTAGE	CUMULATIVE REMAINDER	MULTIPLE OF MEAN	DEVIATION FROM MEAN
0	0	.00	.0	100.0	.000	-2.522
500	2	.40	.4	99.6	.063	-2.363
1000	3	.60	1.0	99.0	.126	-2.204
1500	6	1.21	2.2	97.8	.189	-2.045
2000	8	1.61	3.8	96.2	.252	-1.886
2500	4	.80	4.6	95.4	.315	-1.727

3000	8	1.61	6.2	93.8	.378	-1.568
3500	13	2.62	8.9	91.1	.442	-1.409
4000	6	1.21	10.1	89.9	.505	-1.249
4500	20	4.02	14.1	85.9	.568	-1.090
5000	18	3.62	17.7	82.3	.631	-.931
5500	25	5.03	22.7	77.3	.694	-.772
6000	27	5.43	28.2	71.8	.757	-.613
6500	21	4.23	32.4	67.6	.820	-.454
7000	27	5.43	37.8	62.2	.883	-.295
7500	26	5.23	43.1	54.9	.946	-.136
8000	37	7.44	50.5	49.5	1.009	.023
8500	25	5.03	55.5	49.5	1.072	.182
9000	34	6.84	62.4	37.6	1.135	.341
9500	28	5.63	68.0	32.0	1.198	.500
10000	32	6.44	74.4	25.6	1.262	.660
10500	24	4.83	79.3	20.7	1.325	.819
11000	25	5.03	84.3	15.7	1.388	.978
11500	18	3.62	87.9	12.1	1.451	1.137
12000	19	3.82	91.8	8.2	1.514	1.296
12500	12	2.41	94.2	5.8	1.577	1.455
13000	5	1.01	95.2	4.8	1.640	1.614
13500	6	1.21	96.4	3.6	1.703	1.773
14000	5	1.01	97.4	2.6	1.766	1.932
OVERFLOW	13	2.62	100.0	.0		

TABLE NUMBER 3

ENTRIES IN TABLE

483

MEAN ARGUMENT

19392.867

STANDARD DEVIATION

3435.608

NON-WEIGHTED

UPPER LIMIT	OBSERVED FREQUENCY	PERCENT OF TOTAL	CUMULATIVE PERCENTAGE	CUMULATIVE REMAINDER	MULTIPLE OF MEAN	DEVIATION FROM MEAN
9000	0	.00	.0	100.0	.464	-3.025
9500	0	.00	.0	100.0	.490	-2.880
10000	0	.00	.0	100.0	.516	-2.734
10500	0	.00	.0	100.0	.541	-2.588
11000	0	.00	.0	100.0	.567	-2.443
11500	1	.21	.2	99.8	.593	-2.297
12000	3	.62	.8	99.2	.619	-2.152
12500	2	.41	1.2	98.8	.645	-2.006
13000	6	1.24	2.5	97.5	.670	-1.861
13500	5	1.04	3.5	96.5	.696	-1.715
14000	8	1.66	5.2	94.8	.722	-1.570
14500	15	3.11	8.3	91.7	.748	-1.424
15000	8	1.66	9.9	90.1	.773	-1.279
15500	15	3.11	13.0	87.0	.799	-1.133
16000	12	2.48	15.5	84.5	.825	-.988
16500	18	3.73	19.3	80.7	.851	-.842
17000	22	4.55	23.8	76.2	.877	-.696
17500	26	5.38	29.2	70.8	.902	-.551
18000	21	4.35	33.5	66.5	.928	-.405
18500	20	4.14	37.7	62.3	.954	-.260
19000	42	8.70	46.4	53.6	.980	-.114
19500	24	4.97	51.3	48.7	1.006	.031

UPPER LIMIT	OBSERVED FREQUENCY	MEAN ARGUMENT 1207.008	PERCENT OF TOTAL	CUMULATIVE PERCENTAGE	STANDARD DEVIATION 225.094	CUMULATIVE REMAINDER	MULTIPLE OF MEAN	DEVIATION FROM MEAN
20000	35	7.25	.00	.0	58.6	41.4	1.031	.177
20500	17	3.52	.00	.0	62.1	37.9	1.057	.322
21000	31	6.42	.00	.0	68.5	31.5	1.083	.468
21500	31	6.42	.00	.0	74.9	25.1	1.109	.613
22000	32	6.63	.00	.0	81.6	18.4	1.134	.759
22500	17	3.52	.00	.0	85.1	14.9	1.160	.904
23000	14	2.90	.00	.0	88.0	12.0	1.186	1.050
OVERFLOW	58	12.01	.00	100.0		.0		

TABLE NUMBER 4

ENTRIES IN TABLE		MEAN ARGUMENT 496	PERCENT OF TOTAL	CUMULATIVE PERCENTAGE	STANDARD DEVIATION 225.094	CUMULATIVE REMAINDER	MULTIPLE OF MEAN	DEVIATION FROM MEAN
UPPER LIMIT	OBSERVED FREQUENCY	1207.008						
0	0	.00	.00	.0		100.0	.000	-5.362
100	0	.00	.00	.0		100.0	.083	-4.918
200	0	.00	.00	.0		100.0	.166	-4.474
300	1	.20	.00	.2		99.8	.249	-4.029
400	0	.00	.00	.2		99.8	.331	-3.585
500	2	.40	.00	.6		99.4	.414	-3.141
600	2	.40	.00	1.0		99.0	.497	-2.697
700	5	1.01	.00	2.0		98.0	.580	-2.252
800	10	2.02	.00	4.0		96.0	.663	-1.808
900	25	5.04	.00	9.1		90.9	.746	-1.364
1000	38	7.66	.00	16.7		83.3	.828	-.920
1100	58	11.69	.00	28.4		71.6	.911	-.475
1200	89	17.94	.00	46.4		53.6	.994	-.031
1300	97	19.56	.00	65.9		34.1	1.077	.413
1400	80	16.13	.00	82.1		17.9	1.160	.857
1500	43	8.67	.00	90.7		9.3	1.243	1.302
1600	30	6.05	.00	96.8		3.2	1.326	1.746
1700	12	2.42	.00	99.2		.8	1.408	2.190
1800	4	.81	.00	100.0		.0	1.491	2.634

REMAINING FREQUENCIES ARE ALL ZERO

TABLE NUMBER 5

ENTRIES IN TABLE		MEAN ARGUMENT 681.625	PERCENT OF TOTAL	CUMULATIVE PERCENTAGE	STANDARD DEVIATION 246.855	CUMULATIVE REMAINDER	MULTIPLE OF MEAN	DEVIATION FROM MEAN
UPPER LIMIT	OBSERVED FREQUENCY	681.625						
0	0	.00	.00	.0		100.0	.000	-2.761
100	0	.00	.00	.0		100.0	.147	-2.356
200	0	.00	.00	.0		100.0	.293	-1.951
300	1	12.50	.00	12.5		87.5	.440	-1.546
400	0	.00	.00	12.5		87.5	.587	-1.141
500	0	.00	.00	12.5		87.5	.734	-.736
600	3	37.50	.00	50.0		50.0	.880	-.331
700	0	.00	.00	50.0		50.0	1.027	.074

800	0	.00	50.0	50.0	1.174	.480
900	0	.00	50.0	50.0	1.320	.885
1000	4	50.00	100.0	.0	1.467	1.290

REMAINING FREQUENCIES ARE ALL ZERO

TABLE NUMBER 6		ENTRIES IN TABLE		MEAN ARGUMENT		STANDARD DEVIATION		NON-WEIGHTED	
		497		9.835		82.668			
UPPER LIMIT	OBSERVED FREQUENCY	PERCENT OF TOTAL	CUMULATIVE PERCENTAGE	CUMULATIVE REMAINDER	MULTIPLE OF MEAN	DEVIATION FROM MEAN			
0	489	98.39	98.4	1.6	.000	-.119			
100	0	.00	98.4	1.6	10.168	1.091			
200	0	.00	98.4	1.6	20.336	2.300			
300	2	.40	98.8	1.2	30.503	3.510			
400	0	.00	98.8	1.2	40.671	4.720			
500	0	.00	98.8	1.2	50.839	5.929			
600	2	.40	99.2	.8	61.007	7.139			
700	0	.00	99.2	.8	71.174	8.349			
800	2	.40	99.6	.4	81.342	9.558			
900	1	.20	99.8	.2	91.510	10.768			
1000	1	.20	100.0	.0	101.678	11.978			
REMAINING FREQUENCIES ARE ALL ZERO									

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TABLE NUMBER 8		ENTRIES IN TABLE		MEAN ARGUMENT		STANDARD DEVIATION		NON-WEIGHTED	
7		731.000		246.165					
UPPER LIMIT	OBSERVED FREQUENCY	PERCENT OF TOTAL	CUMULATIVE PERCENTAGE	CUMULATIVE REMAINDER	MULTIPLE OF MEAN	DEVIATION FROM MEAN			
0	0	.00	.0	100.0	.000	-2.970			
100	0	.00	.0	100.0	.137	-2.563			
200	0	.00	.0	100.0	.274	-2.167			
300	0	.00	.0	100.0	.410	-1.751			
400	1	14.29	14.3	85.7	.547	-1.345			
500	0	.00	14.3	85.7	.684	-.938			
600	3	42.86	57.1	42.9	.821	-.532			
700	0	.00	57.1	42.9	.958	-.126			
800	0	.00	57.1	42.9	1.094	.280			
900	0	.00	57.1	42.9	1.231	.687			
1000	3	42.86	100.0	.0	1.368	1.093			
REMAINING FREQUENCIES ARE ALL ZERO									

TABLE NUMBER 9		ENTRIES IN TABLE		MEAN ARGUMENT		STANDARD DEVIATION		NON-WEIGHTED	
		484		1.014		.119			

UPPER LIMIT	OBSERVED FREQUENCY	PERCENT OF TOTAL	CUMULATIVE PERCENTAGE	CUMULATIVE REMAINDER	MULTIPLE OF MEAN	DEVIATION FROM MEAN
0	0	.00	.0	100.0	.000	-8.497
1	477	90.55	98.6	1.4	.986	-.121
2	7	1.45	100.0	.0	1.971	8.255

REMAINING FREQUENCIES ARE ALL ZERO

TABLE NUMBER 10

ENTRIES IN TABLE		MEAN ARGUMENT	STANDARD DEVIATION		NON-WEIGHTED	
484		5325.473	125.264			
UPPER LIMIT	OBSERVED FREQUENCY	PERCENT OF TOTAL	CUMULATIVE PERCENTAGE	CUMULATIVE REMAINDER	MULTIPLE OF MEAN	DEVIATION FROM MEAN
0	0	.00	.0	100.0	.000	-42.514
500	0	.00	.0	100.0	.094	-38.522
1000	0	.00	.0	100.0	.188	-34.531
1500	0	.00	.0	100.0	.282	-30.539
2000	0	.00	.0	100.0	.376	-26.548
2500	0	.00	.0	100.0	.469	-22.556
3000	0	.00	.0	100.0	.563	-18.565
3500	0	.00	.0	100.0	.657	-14.573
4000	0	.00	.0	100.0	.751	-10.581
4500	1	.21	.2	99.8	.845	-6.590
5000	5	1.03	1.2	98.8	.939	-2.598
5500	471	97.31	98.6	1.4	1.033	1.393
6000	4	.83	99.4	.6	1.127	5.385
6500	3	.62	100.0	.0	1.221	9.376

REMAINING FREQUENCIES ARE ALL ZERO

TABLE NUMBER 21

ENTRIES IN TABLE		MEAN ARGUMENT	STANDARD DEVIATION		NON-WEIGHTED	
482		5872.456	498.013			
UPPER LIMIT	OBSERVED FREQUENCY	PERCENT OF TOTAL	CUMULATIVE PERCENTAGE	CUMULATIVE REMAINDER	MULTIPLE OF MEAN	DEVIATION FROM MEAN
5000	0	.00	.0	100.0	.851	-1.752
5500	60	12.45	12.4	87.6	.937	-.748
6000	294	61.00	73.4	26.6	1.022	.256
6500	73	15.15	88.6	11.4	1.107	1.260
7000	35	7.26	95.9	4.1	1.192	2.264
7500	12	2.49	98.3	1.7	1.277	3.268
8000	5	1.04	99.4	.6	1.362	4.272
8500	3	.62	100.0	.0	1.447	5.276

REMAINING FREQUENCIES ARE ALL ZERO

TABLE NUMBER 22

ENTRIES IN TABLE MEAN ARGUMENT STANDARD DEVIATION

497	9673.145	3591.861	NON-WEIGHTED		
UPPER LIMIT	OBSERVED FREQUENCY	PERCENT OF TOTAL	CUMULATIVE PERCENTAGE	CUMULATIVE REMAINDER	MULTIPLE OF MEAN
0	1	.20	.2	99.8	.000
500	0	.00	.2	99.8	.052
1000	0	.00	.2	99.8	.103
1500	2	.40	.6	99.4	.155
2000	1	.20	.8	99.2	.207
2500	4	.80	1.6	98.4	.258
3000	7	1.41	3.0	97.0	.310
3500	5	1.01	4.0	96.0	.362
4000	7	1.41	5.4	94.6	.414
4500	9	1.81	7.2	92.8	.465
5000	12	2.41	9.7	90.3	.517
5500	12	2.41	12.1	87.9	.569
6000	17	3.42	15.5	84.5	.620
6500	21	4.23	19.7	80.3	.672
7000	25	5.03	24.7	75.3	.724
7500	14	2.82	27.6	72.4	.775
8000	28	5.63	33.2	66.8	.827
8500	23	4.63	37.8	62.2	.879
9000	26	5.23	43.1	56.9	.930
9500	18	3.62	46.7	53.3	.982
10000	35	7.04	53.7	46.3	1.034
10500	29	5.84	59.6	40.4	1.085
11000	28	5.63	65.2	34.8	1.137
11500	26	5.23	70.4	29.6	1.189
12000	22	4.43	74.8	25.2	1.241
12500	16	3.22	78.1	21.9	1.292
13000	24	4.83	82.9	17.1	1.344
13500	14	2.82	85.7	14.3	1.396
14000	18	3.62	89.3	10.7	1.447
OVERFLOW	53	10.66	100.0	.0	
					DEVIATION FROM MEAN
					-2.693
					-2.554
					-2.415
					-2.275
					-2.136
					-1.997
					-1.858
					-1.719
					-1.579
					-1.440
					-1.301
					-1.162
					-1.023
					-.883
					-.744
					-.605
					-.466
					-.327
					-.187
					-.048
					.091
					.230
					.369
					.509
					.648
					.787
					.926
					1.065
					1.205

TABLE NUMBER 23

ENTRIES IN TABLE	MEAN ARGUMENT	STANDARD DEVIATION	NON-WEIGHTED		
482	20915.421	3612.657			
UPPER LIMIT	OBSERVED FREQUENCY	PERCENT OF TOTAL	CUMULATIVE PERCENTAGE	CUMULATIVE REMAINDER	MULTIPLE OF MEAN
9000	0	.00	.0	100.0	.430
9500	0	.00	.0	100.0	.454
10000	0	.00	.0	100.0	.478
10500	0	.00	.0	100.0	.502
11000	1	.21	.2	99.8	.526
11500	0	.00	.2	99.8	.550
12000	0	.00	.2	99.8	.574
12500	2	.41	.6	99.4	.598
13000	3	.62	1.2	98.8	.622
13500	4	.83	2.1	97.9	.645
14000	4	.83	2.9	97.1	.669
14500	6	1.24	4.1	95.9	.693
					DEVIATION FROM MEAN
					-3.298
					-3.160
					-3.021
					-2.883
					-2.745
					-2.606
					-2.468
					-2.329
					-2.191
					-2.053
					-1.914
					-1.776

15000	4	.83	5.0	95.0	.717	-1.637
15500	7	1.45	6.4	93.6	.741	-1.499
16000	7	1.45	7.9	92.1	.765	-1.361
16500	14	2.90	10.8	89.2	.789	-1.222
17000	15	3.11	13.9	86.1	.813	-1.084
17500	17	3.53	17.4	82.6	.837	-.945
18000	20	4.15	21.6	78.4	.861	-.807
18500	17	3.53	25.1	74.9	.885	-.669
19000	23	4.77	29.9	70.1	.908	-.530
19500	25	5.19	35.1	64.9	.932	-.392
20000	25	5.19	40.2	59.8	.956	-.253
20500	22	4.56	44.8	55.2	.980	-.115
21000	29	6.02	50.8	49.2	1.004	.023
21500	29	6.02	56.8	43.2	1.028	.162
22000	22	4.56	61.4	38.6	1.052	.300
22500	29	6.02	67.4	32.6	1.076	.439
23000	21	4.36	71.8	28.2	1.100	.577
OVERFLOW	136	28.22	100.0	.0		

FUTURE RANDOM NUMBER SEED IS (OCTAL) 277404621561

APPENDIX E. SAMPLE CONFIDENCE INTERVAL OPTION OUTPUT

DATE 032372 PAGE 06

TABLE III. STATISTICAL RESULTS

CONFIDENCE INTERVALS ON REQUESTED STATISTICS

WARNING - CONFIDENCE INTERVALS WITH LIMITS OF 0 OR 100 ARE PROBABLY
NOT 99 PERCENT CONFIDENCE INTERVALS

STATISTICS BASED ON 8 OBSERVATIONS.

99 PERCENT CONFIDENCE INTERVALS		
LOWER LIMIT	MEAN	UPPER LIMIT

STATISTICAL CONFIDENCE INTERVALS FOR 500 SIMULATIONS.

THE PERCENTAGE OF SUCCESSFUL LAUNCH REQUESTS

THE PERCENTAGE OF UNSUCCESSFUL LAUNCH REQUESTS

THE PERCENTAGE OF VEHICLES LAUNCHED AT A SUBSEQUENT WINDOW

REASONS FOR REQUEST FAILURES:

BOOSTER UNAVAILABILITY

ORBITER UNAVAILABILITY

BOTH STAGES UNAVAILABLE

LUT UNAVAILABILITY

PAD UNAVAILABILITY

FAILURES AND REMOVAL FROM PAD

F-2

THE FOLLOWING ARE THE USER REQUESTED STATISTICS

TIME FROM LANDING TO STORAGE (BOOSTER)

ORBITER SAFING FACILITY UTILIZATION (PERCENT)

LAUNCH PAD 1 UTILIZATION (PERCENT)

BOOSTER SAFING QUEUE TIME

ORBITER MAINTENANCE QUEUE TIME

APPENDIX F. EXPLANATION OF INDIVIDUAL CONTROL CARDS

This appendix gives a brief operational description of the various control cards contained in the example deck setups shown in the document. Further information about these control statements is given in the MSFC Program Procedures Manual with the page numbers in parentheses. (Refer to Figure F-1 for the correct column positioning).

1. @ADD MODEL.

This card will add the contents of file MODEL to the run stream. This card must follow the @XQT GPSSE8. MAPGPS card for each execution of the GPSS program. (6.3.1-80)

2. @ASG, T NAME, F2

This control card assigns a temporary file location called NAME on a magnetic drum unit. The F2 represents the type of drum selected for mass storage. Note that an @ASG card must be used for each file referenced within the run. (6.3.1-29)

3. @ASG, T TAPENAME, T, SAVEXX

This card creates a tape called TAPENAME and is maintained in the MSFC tape library for the time specified by the code XX. The code XX represents the following:
(Technical Bulletin No. 17)

02--14 day retention
03--30 day retention
04--90 day retention
05--6 month retention

4. @ASG, T TAPENAME, T, 12345

This card assigns the use of a tape drive to the run. In this example, the tape file is called TAPENAME and reel number 12345 is loaded onto the drive unit. (6.3.1-33)

5. @BRKPT PUNCH\$

This card causes the current normal punch file to be closed and queued for punching and starts a new normal punch file. (6.3.1-101)

6. @BRKPT PUNCH\$/FN
This statement causes the current normal punch file to be closed and queued for printing and routes all future output of the normal punch type to a file whose name is FN. The file whose name is FN must be currently assigned to the run. (6.3.1-102)
7. @COPIN TAPENAME, GPSSE8.
This card causes the transfer of the program instruction from the tape called TAPENAME, to a file name GPSSE8. (6.3.1-147)
8. @COPIN TAPENAME, TPF\$.
This card causes the transfer of the program instructions from the tape, called TAPENAME, to a temporary program file. (6.3.1-148)
9. @COPOUT GPSSE8, TAPENAME
This card is a reverse operation of the @COPIN card. The program instructions are transferred from the file GPSSE8 to the tape named TAPENAME. (6.3.1-146)
10. @COPY, G TAPENAME, MODEL.
This control card causes the transfer of a data file (the GPSS System Model Description) from the tape named TAPENAME to a drum location named MODEL. (6.3.1-144)
11. @COPY, GM MODEL, TAPENAME
This represents the reverse operation of the @COPY, G TAPENAME, MODEL. command. (6.3.1-144)
12. @COPY, RSA MASTER*GPSSE8., GPSSE8.
This control card is used to command execution from a drum file. This card copies from the drum a program called MASTER*GPSSE8 into temporary location named GPSSE8. (6.3.1-144)
13. @DATA FILE2, MODEL
This control card permits the updating of the GPSS model network contained on a file named FILE2. The new revision is stored in a file named MODEL (corrections or changes to MODEL follow this control card and FILE2 remains unchanged). (6.3.1-97)
14. @DATA, L DUMMYP, MODEL
This control allows the revision of the DUMMYP file and creates an updated file named MODEL. The revisions to DUMMYP follow this control card. A listing of the data in MODEL is then generated. (6.3.1-97).

15. @END
This control card marks the end of the data corrections following the DATA control card. (6.3.1-98)
16. @ERS GPSSE8.
This control card will erase all contents of the program file named GPSSE8. (This card is used in decks that generate a new program file.) (6.3.1-156)
17. @FIN
Signifies the end of the control deck. (This is the last card on all deck setups.) (6.3.1-80)
18. @FOR, US GPSSE8.NAME, GPSSE8.NAME
Control card used to include a new or modified Fortran user written subroutine in the GPSS run stream. The word NAME should be replaced on the control card with the title of the subroutine. The U option allows modification of individual cards in the subroutine. An I option allows the inclusion of an entire subroutine package but not the updating of individual cards (6.3.1-103)
19. @FREE TAPENAME
Releases the tape called TAPENAME. (In cases where the information on the tape has been copied onto the drums, the tape drive unit is no longer needed and, therefore, released from the run.) (6.3.1-88)
20. @HDG, P CASE STUDY TITLE
This heading card prints any desired title on the top of each page of the printout. The page number and data are also printed. All columns on the card can be used. In this case, the title CASE STUDY TITLE is printed on top of each page of output. (6.3.1-124)
21. @MAP, LX GPSSE8.MAP, GPSSE8.MAPGPS
This card causes the collection of all program elements into the program instruction element by the name of MAPGPS. (6.6-4)
22. @MOVE TAPENAME, 2
This control card is used for positioning the tape called TAPENAME at the beginning of the selected file. (The example indicates the movement of the tape past 2 files from its present position.) (6.3.1-137)
23. @PREP GPSSE8.
This command is used to prepare a program file on FASTRAND for subsequent referencing as a library by the collector. This card must always be used before a @MAP control card. (6.3.1-153)

24. @PRT, T
This card produces a table of contents of the temporary program file (TPF\$). (6.3.1-157)
25. @PRT, T GPSSE8.
This card will generate a table of contents of all elements contained in the program file GPSSE8. (6.3.1-151)
26. @REWIND TAPENAME
Executive command which causes the rewinding of the tape drive called TAPENAME. Execution of this control causes the tape to be positioned at the starting point of the first tape file. (6.3.1-138)
27. @RUN, //P SAMPLE, 999999, PORTERBIN225, 3, 150
The run is always the first control card in the 1108 deck setup. This card contains all the information used by the computer for accounting and identification purposes. The three options available on the RUN card are P (production), T (development, and A (maintenance and checkout). SAMPLE1 is the run I.D. code assigned to the particular deck for use by the computer, 999999 is the accounting job number, PORTER is the programmer's name in six characters, and 225 identifies the programmer's BIN location. The 3 is the maximum CPU time (min) for the run. The 150 represents the maximum number of pages of output produced by the program. (Technical Bulletin No. 16)
28. @SETC 2
Sets the computer condition word to the value of 2 and causes the generation of a punched deck of the absolute GPSS model. (6.3.1-118)
29. @SETC 3
Sets the computer condition word to the value of 3, thus eliminating the listing of the model network. A separate header page is generated for each case to be executed. (6.3.1-118)
30. @SETC 4
Sets the computer condition word to the value of 4, thus causing a printout indicating the distributions and their associated parameters from the GPSS model that were defined for each processing operation. (An extra control card, @ADD MODEL, must be inserted into the program deck when this control option is used.) (6.3.1-118)

31. @SETC 5
Sets the computer condition word to the value of 5, thus causing the System Model to be read for headings used in the collection of confidence interval statistics. (An extra control card, @ADD MODEL, must be inserted into the program deck when this control option is used.)
(6.3.1-118)
32. @XQT GPSSE8.MAPGPS
Control card command indicating the execution of the GPSS program instructions called MAPGPS contained on a file named GPSSE8. (6.3.1-108)
33. @XQT,LA SYS\$*MSFC\$.LISTIT
This card executes a special program that generates a listing of the contents of the temporary program file.
(6.3.1-106)

GENERAL CARD DESIGN									
26	@REWIND	TAPENAME							
27	@RUN, //	P SAMPLE, 999999, PORTERBIN 225, 3, 150							
28	@SETC	2							
29	@SETC	3							
30	@SETC	4							
31	@SETC	5							
32	@XQT	GPSSE8.MAPGPS							
33	@XQT, LA	SYS\$*MSFC\$. LISTIT							

MSFC-MSA, AS

FIGURE F-1. EXAMPLE CONTROL CARD FORMAT (Sheet 2 of 2)

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