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FRICTION, WEAR, AND THERMAL STABILITY STUDIES OF SOME ORGANOTIN AND ORGANOSILICON COMPOUNDS

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FRICTION, WEAR, AND THERMAL STABILITY STUDIES OF SOME ORGANOTIN AND ORGANOSILICON COMPOUNDS

by William R. Jones, Jr.

Lewis Research Center

SUMMARY

Thermal decomposition temperatures were determined for a number of organotin and organosilicon compounds. A ball-on-disk sliding friction apparatus was used to determine the friction and wear characteristics of two representative compounds, (1) 3-tri-n-butylstannyl (diphenyl) and (2) 3-tri-n-butylsilyl (diphenyl). Results were compared to those previously obtained with a polyphenyl ether and a C-ether. Friction and wear test conditions included a 1-kilogram load, a 17-meter-per-minute (100-rpm) sliding speed, 25° to 225° C disk temperatures, a dry-air (<100 ppm water) atmosphere, and a 25-minute test duration. In general, the 3-tri-n-butylstannyl (diphenyl) and the 3-tri-n-butylsilyl (diphenyl) compounds yielded friction and wear results either lower than or similar to those obtained with a polyphenyl ether and a C-ether.

The maximum thermal decomposition temperatures obtained in the silicon and tin series were 358° C with 3-tri-n-octylsilyl (diphenyl) and 297° C with 4-triphenylstannyl (diphenyl). Increasing the steric hindrance around the silicon or tin atoms increased the thermal stability. Future work with these compounds will emphasize their possible use as antiwear additives rather than base fluids, because of preparation difficulties, high cost, and lower than expected thermal decomposition temperatures.

INTRODUCTION

Higher flight speeds and higher output engines of advanced aircraft are creating needs for improved or new high-temperature liquid lubricants (ref. 1). Several classes of synthetic fluids have been proposed as potential lubricants for high-temperature applications. These classes include polyphenyl ethers (refs. 2 to 6), C-ethers (refs. 7 to 10), polyethers (refs. 11 to 13), and synthetic hydrocarbons (refs. 12 to 14). Another class of compounds which has elicited considerable interest as possible high-temperature lubricants or lubricant additives is the organometallics. This group includes siloxanes, ferrocenes, silanes, and stannanes (refs. 15 to 19).

Tetraalkylsilanes have been studied as base stocks for high-temperature lubricant formulations (ref. 17). Thermal decomposition temperatures as high as 350° C were reported. Some of these fluids exhibited good viscosity properties up to 316° C (i.e., they have a wide liquid range). They also exhibited good lubricating ability to 200° C in four ball tests. The major difficulty encountered with these fluids was their poor stability in oxidation and corrosion tests at 260° C.

A number of organotin compounds have been synthesized for use as lubricant additives to improve the oxidation stability (ref. 18) or the antiwear behavior (ref. 19) of base stocks. Recently, Zimmer and coworkers (refs. 20 to 23) have prepared a variety of sterically hindered Group IVA organometallic compounds. These compounds are characterized by decreased reactivity because of the bulky groups shielding the central atom.

The objectives of this investigation were (1) to determine the thermal decomposition temperatures (by isoteniscope) for a number of organotin and organosilicon compounds, (2) to determine the friction and wear characteristics of a selected organotin and organosilicon compound, and (3) to compare these results to those obtained with a polyphenyl ether and a C-ether.

Experimental conditions with the ball-on-disk sliding friction apparatus included a 1-kilogram load (initial Hertz stress, 10^9 N/m^2), a 17-meter-per-minute (100-rpm) sliding speed, 25° to 225° C disk temperatures, a dry-air (<100 ppm water) atmosphere, and a 25-minute test duration. All organotin and organosilicon compounds were synthesized at the University of Cincinnati, Cincinnati, Ohio, under NASA grant NGL 36-004-056.

EXPERIMENTAL LUBRICANTS

The general chemical structures of the organotin and organosilicon compounds are shown in figure 1. The two compounds selected for the friction and wear tests, a 3-tri-n-butylstannyl (diphenyl) and 3-tri-n-butylsilyl (diphenyl), are illustrated by figures 1(b) and (d), respectively, where R is the n-butyl alkyl group.

The two reference fluids used in these studies were a five-ring polyphenyl ether and a C-ether. The polyphenyl ether is an isomeric mixture of five-ring components and contains no additives. The C-ether is a blend of three- and four-ring components which are structurally related to the polyphenyl ethers. This is a base fluid which con-

tains only an antifoam additive. Some typical properties of these four fluids are listed in table I.

APPARATUS

Ball-on-Disk Apparatus

The ball-on-disk sliding friction apparatus is shown in figure 2. The test specimens were contained inside a stainless-steel chamber. The atmosphere was controlled with respect to moisture content.

A stationary 0.476-centimeter-radius ball was placed in sliding contact with a rotating 6.3-centimeter-diameter disk. A sliding speed of 17 meters per minute (100 rpm) was maintained. A normal load of 1 kilogram (initial Hertz stress, 10^9 N/m^2) was applied with a deadweight. Balls and disks were made of CVM M-50 steel having a Rockwell C hardness of 62 to 64.

The disk was partially submerged in a polyimide cup containing the test lubricant and was heated by induction. Bulk lubricant temperature was measured with a thermocouple. Disk temperature was monitored with an infrared pyrometer. Frictional force was measured with a strain gage and was recorded on a strip-chart recorder.

Thermal Decomposition Apparatus (Tensimeter)

The tensimeter (an automated isoteniscope) apparatus shown in figure 3 consists of two units: the sample oven with its associated temperature programmer and the electronics module containing the recorder. The basic operation of the tensimeter consists of heating a liquid sample and of plotting the isothermal rate of increase of vapor pressure (due to thermal decomposition) at a number of temperatures.

The sample assembly appears in figure 4. The sample cell is a glass bulb having a 5-cubic-centimeter volume, with a stem extending through the oven wall to a valve and pressure transducer mounted outside the oven.

Atmosphere Monitoring and Control

Dry air (<100 ppm water) was the only atmosphere used in this investigation. It was obtained by drying and filtering service air. The moisture content was monitored by a moisture analyzer with an accuracy of ± 10 parts per million.

PROCEDURE

Friction and Wear Tests

Disks were ground and lapped to a root-mean-square surface finish of approximately 10×10^{-8} to 20×10^{-8} meter (4 to 8 µin.). The root-mean-square surface finish of the balls was about 2.5×10⁻⁸ meter (1 µin.). Specimens were scrubbed with a paste of levigated alumina and water, rinsed with tap water and distilled water, and then placed in a desiccator.

The organotin and organosilicon compounds were tested in the as-received condition. The specimens were mounted, and 20 cubic centimeters of lubricant were placed in the lubricant cup. The test chamber $(3.7 \times 10^3 \text{ cm}^3 \text{ volume})$ was purged with dry air (<100 ppm water) for 10 minutes at a flow rate in excess of 50×10^3 cubic centimeters per hour. The disk was heated by induction to the test temperature while rotating, and the ball was then loaded against the disk. The test atmosphere flow rate was reduced to 35×10^3 cubic centimeters per hour, and a 6.9×10^{-3} -newton-per-square-meter (1-psig) pressure was maintained in the chamber. The lubricant was heated only by heat transfer from the rotating disk. The bulk lubricant temperature was essentially the same as the disk. The bulk lubricant temperature was essentially the same as the disk. The bulk lubricant temperature age.

Frictional force and bulk lubricant temperature were continuously recorded. Disk temperature was continuously monitored. Experiments were terminated after 25 minutes, and the rider (ball) wear scar was recorded.

Thermal Decomposition Tests

A 3- to 4-cubic-centimeter sample of test fluid is placed in the sample cell. The cell is attached to the cell assembly, and the fluid is degassed and refluxed under vacuum. The cell assembly is then placed in the oven. The sample is heated to an initial temperature about 50° C (122° F) below the suspected decomposition temperature. After a 5-minute stabilization period, the increase in vapor pressure, if any, is recorded as a vertical bar during a fixed time interval. The programmer raises the sample temperature by a preset amount, usually 5° C, and the previous process is repeated. A typical plot appears in figure 5.

This is essentially a plot of the logarithm of the isothermal rate of vapor pressure increase as a function of reciprocal absolute temperature. A straight line is drawn which connects the tops of the recorded bars at the higher temperatures. The intersection with the temperature reference axis is the thermal decomposition temperature

 T_D . This temperature axis corresponds to a pressure rise of 50 torr per hour, which is the definition of the T_D . The activation energy for decomposition E' may be calculated from the slope of the straight line.

RESULTS AND DISCUSSION

Thermal Decomposition Results

Table II lists the thermal decomposition results for both the organotin and organosilicon series of compounds. Both meta- and paralinked compounds have been included, and their general chemical structures are illustrated in figure 1.

Emphasis was placed on the metalinked compounds for two reasons. First, from theoretical considerations, the metalinked compounds should be more thermally stable than their para-analogs. Thermal decomposition usually proceeds by means of free radicals. If the potential free radical cannot be stabilized by resonance, it will be more difficult to form than one which is stabilized. This stabilization is not possible with metalinkages. Second, metalinkages produce a more asymmetrical molecule, which should reduce the melting points and thus produce a greater liquid range.

A minimum quantity of 3 to 5 cubic centimeters of fluid is required for each thermal decomposition test. If a compound was a solid at room temperature or was available only in small quantities, a mixing procedure had to be utilized. The compound was mixed with a four-ring polyphenyl ether to produce a concentration of 20 to 50 percent by volume. The polyphenyl ether had a thermal decomposition temperature (T_D) of about 450° C. Therefore, any decomposition taking place below this temperature must be attributed to the tin or silicon compound. However, there is a dilution effect which causes a small error (approx 7 percent) in the T_D measurement. The amount of the error was determined by checking the T_D of a standard fluid mixed with the polyphenyl ether. The fluids tested as mixtures are noted in table II and their corrected T_D values are given. In general, the reproducibility of T_D values for pure compounds is better than $\pm 5^{\circ}$ C. However, the T_D values for the mixtures may vary by $\pm 10^{\circ}$ C.

The substituents in table II are listed in order of increasing steric hindrance. As can be seen, the thermal stability generally increases with increasing steric hindrance (i.e., increasing size of the substituent groups). This is somewhat surprising, since steric crowding usually causes a decrease in thermal stability (ref. 24). A possible explanation could be that if vibrational energies are evenly distributed over the entire molecule, a higher energy (i.e., higher temperature) would be required to rupture the same bond in a larger molecule than in a smaller one.

It is apparent that the steric factor does not influence the T_D in a linear fashion. Much greater changes in T_D occur in the latter part of the series (<u>n</u>-hexyl to neopentyl) than in the initial part (methyl to <u>n</u>-hexyl). An example from the literature may serve to illustrate this nonlinear effect on reaction rates. Dostrovsky and Hughes (refs. 25 to 29) have shown that the rate of basic ethanolysis of alkyl bromides decreases enormously with the increasing bulk of groups attached to the carbon atom under attack. The reaction is

$$C_{2}H_{5}OH + RBr \rightarrow Br + product$$
 (1)

Table III shows the relative reaction rates as the size of the R group increases. Again the greatest effect occurs with the bulky neopentyl group.

A second interesting observation from table Π is that the paracompounds do not appear to be less stable than their meta-analogs. Within experimental error their thermal stabilities appear to be essentially the same. Apparently resonance stabilization of decomposition products is not an important factor with the compounds studied in this investigation.

The highest thermal stability attained in the tin series was 297° C with the paraphenyl derivative. In the silicon series, the meta-n-octyl derivative yielded the highest T_D (358° C). One disubstituted compound (3, 3'bis(tri-n-propyl-tin)biphenyl) was synthesized and found to be less thermally stable ($T_D = 214^{\circ}$ C) than the monosubstituted derivative ($T_D = 236^{\circ}$ C).

Friction and Wear Results

The meta-n-butyl derivatives were chosen as the representative compounds from each series for the friction and wear tests. The n-butyl derivatives were selected because of their ease of preparation and the fact that they were both viscous liquids at room temperature.

Figure 6 contains the friction and wear results for these two fluids. The temperature range studied was 25° to 225° C. The upper temperature limit was dictated by fluid volatility. As can be seen, the tin derivative yields slightly lower wear than the silicon analog at low temperatures (25° to 75° C) but greater wear at the higher temperatures (75° to 225° C). The increase in wear rate at the high temperatures for both fluids may in part be due to loss of lubricant by evaporation.

The coefficient of friction for the tin derivative is essentially constant over the entire temperature range. The silicon analog exhibited a linearly increasing friction coefficient.

A comparison of the tin and silicon results with those obtained previously with two high-temperature aromatic lubricants, a C-ether (ref. 9) and a five-ring polyphenyl

ether (5P-4E) (ref. 10), appears in figure 7. As can be seen, both the tin and silicon compounds yielded lower wear than the 5P-4E over practically the entire overlapping temperature range. From 25° to 75° C the tin and silicon compounds exhibit wear rates similar to those obtained with a C-ether. From 75° to 225° C, the tin compound yielded wear similar to that for the C-ether, and the silicon compound yielded wear lower than that for the C-ether. Figure 8 summarizes the wear results for the four fluids at four disk temperatures.

The coefficient of friction as a function of temperature for all four fluids appears in figure 7. In general, all fluids exhibited similar friction coefficients from 25° to 150° C. From 150° to 225° C, the lowest friction was observed with the tin compound ' and the highest with 5P-4E.

It must be emphasized that this study is only a preliminary investigation of the friction and wear characteristics of the tin and silicon compounds. Because of the difficulty in preparation, high cost, and lower than expected thermal decomposition temperatures of these compounds, future work will emphasize their possible use as anti-wear additives rather than as base fluids.

SUMMARY OF RESULTS

Thermal decomposition temperatures were determined for a number of organotin and organosilicon compounds. A ball-on-disk sliding friction apparatus was used to determine the friction and wear characteristics of two representative compounds, (1) 3-tri-n-butylstannyl (diphenyl) and (2) 3-tri-n-butylsilyl (diphenyl). Conditions included a dry-air atmosphere (<100 ppm water), a 1-kilogram load, a 17-meter-perminute sliding speed, and disk temperatures of 25° to 225° C. Test specimens were made of CVM M-50 steel. The major results were as follows:

1. In general, the 3-tri-n-butylstannyl (diphenyl) and the 3-tri-n-butylsilyl (diphenyl) compounds yielded friction and wear results either lower than or similar to those obtained with a polyphenyl ether and a C-ether.

2. The maximum thermal decomposition temperatures obtained in the silicon and tin series were 358° and 297° C, respectively.

3. Increasing the steric hindrance (i.e., increasing the size of the substituent groups) around the silicon or tin atoms increased the thermal stability in both series.

4. Future work with these compounds will emphasize their possible use as anti-

wear additives rather than base fluids because of preparation difficulties, high cost, and lower than expected thermal decomposition temperatures.

Lewis Research Center, National Aeronautics and Space Administration, Cleveland, Ohio, December 4, 1972, 501-24.

REFERENCES

- Bisson, Edmond E.; and Anderson, William J.: Advanced Bearing Technology. NASA SP-38, 1964.
- Mahoney, C. L.; Barnum, E. R.; Kerlin, W. W.; and Sax, K. J.: Meta-Linked Polyphenyl Ethers as High-Temperature Radiation-Resistant Lubricants. ASLE Trans., vol. 3, no. 1, Apr. 1960, pp. 83-92.
- Loomis, William R.: Evaluation of Five Bearing-Separator Materials and Polyphenyl Ether Lubricants for Use in Space Power Generation Systems. NASA TN D-2663, 1965.
- Accinelli, J. B.; Borsoff, V. N.; Kerlin, W. W.; and Beaubien, S. J.: Gear and Bearing Lubrication in External Environments with Polyphenyl Ethers. Paper 60-LUB-8, ASME, Oct. 1960.
- Jones, William R., Jr.; Hady, William F.; and Swikert, Max A.: Lubrication with Some Polyphenyl Ethers and Super-Refined Mineral Oils in a 600^o F (316^o C) Inserted Vane Pump Loop. NASA TN D-5096, 1969.
- Jones, William R., Jr.; and Hady, William F.: Effect of Humidity and a Wettability Additive on Polyphenyl Ether Boundary Lubrication of Steel in Air and Nitrogen to 350⁰ C. NASA TN D-6055, 1970.
- McHugh, Kenneth L.; and Stark, Louis R.: Properties of a New Class of Polyaromatics and Use as High-Temperature Lubricants and Functional Fluids. ASLE Trans., vol. 9, no. 1, Jan. 1966, pp. 13-23.
- Johnson, Robert L.; Loomis, William R.; and Ludwig, Lawrence P.: Bearings, Lubricants, and Seals for Actuators and Mechanisms. NASA Space Shuttle Technology Conference. Vol. Π - Structure and Materials. NASA TM X-2273, 1971, pp. 601-632.

- Jones, William R., Jr.; and Hady, William F.: Boundary Lubrication of Formulated C-Ethers in Air to 300⁰ C. I. Phosphorus Ester Additives. NASA TN D-6915, 1972.
- Jones, William R., Jr.: The Effect of Oxygen Concentration on the Boundary Lubricating Characteristics of an Unformulated C-Ether to 300⁰ C. NASA TN D-7155, 1972.
- 11. Gumprecht, William H.: PR-143--A New Class of High-Temperature Fluids. ASLE Trans., vol. 9, no. 1, Jan. 1966, pp. 24-30.
- Parker, Richard J.; Bamberger, Eric N.; and Zaretsky, Erwin V.: Bearing Torque and Fatigue Life Studies with Several Lubricants for Use in the Range 500^o to 700^o F. NASA TN D-3948, 1967.
- Loomis, William R.; Townsend, Dennis P.; and Johnson, Robert L.: Lubricants for Inerted Lubrication Systems in Engines for Advanced Aircraft. NASA TN D-5420, 1969.
- Jones, William R., Jr.; and Hady, William F.: Boundary Lubrication and Thermal Stability Studies with Five Liquid Lubricants in Nitrogen to 400⁰ C. NASA TN D-6251, 1971.
- 15. Gunderson, Reigh C.; and Hart, Andrew W.; eds: Synthetic Lubricants. Reinhold Publ. Corp., 1962.
- 16. Dolle, R. E.: New Candidate High Temperature Lubricants for Advanced Aerospace Systems. Rep. ML-TDR-64-153, Air Force Materials Lab., Aug. 1964. (Available from DDC as AD-449716.)
- Harris, Edward E.; Milnes, Frank J.; Rayner, R. J.; and Dombchewsky, R.: Research and Development on the Synthesis of Tetraalkyl Silanes. Olin Mathieson Chem. Corp. (WADC TR 57-181, AD-151074), Mar. 1958.
- Stemniski, John R.; Wilson, Glenn R.; Smith, John O.; and McHugh, Kenneth L.: Antioxidants for High-Temperature Lubricants. ASLE Trans., vol. 7, no. 1, Jan. 1964, pp. 43-54.
- Smith, John O.; Wilson, Glenn R.; and McElhill, Elizabeth A.: Research on High-Temperature Additives for Lubricants. Monsanto Chem. Co. (ASD-TR-59-191, Part V, AD-400950), Nov. 1962.
- Zimmer, Hans; Hechenbleikner, I.; Homberg, Otto A.; and Danzik, Mitchell: Sterically Hindered Group 1VA Organometallics. Preparation and Properties of Certain Neopentyltins. J. Org. Chem., vol. 29, no. 9, Sept. 1964, pp. 2632-2636.

- Zimmer, Hans; and Homberg, Otto A.: Sterically Hindered Group IV-A Organometallics. V. Preparation and Properties of Certain Neopentyl- and Neophylleads. J. Org. Chem., vol. 31, no. 3, Mar. 1966, pp. 947-949.
- Zimmer, Hans; Holmberg, Otto A.; and Jayawant, Madhusudan: Sterically Hindered Group 1Va Organometallics. VI. Preparation and Some Properties of Neophyltins and Related Compounds. J. Org. Chem., vol. 31, no. 11, Nov. 1966, pp. 3857-3860.
- Zimmer, H.; Bayless, A.; and Christopfel, W.: Sterically Hindered Group 1VA Organometallics. VIII. Preparation and Some Properties of Neohexyltin Compounds. J. Organometallic Chem., vol. 14, 1968, pp. 222-226.
- 24. Blake, Edward S.; Hammann, William C.; Edwards, James W.; Reichard, Thomas E.; and Art, Morris R.: Thermal Stability as a Function of Chemical Structure. J. Chem. Eng. Data, vol. 6, no. 1, Jan. 1961, pp. 87-98.
- 25. Dostrovsky, I.; and Hughes, E. D.: Mechanism of Substitution at a Saturated Carbon Atom. Part XXVI. The Role of Steric Hindrance. Sec. A. Introductory Remarks, and a Kinetic Study of the Reactions of Methyl, Ethyl, n-Propyl, isoButyl, and neoPentyl Bromides with Sodium Ethoxide in Dry Ethyl Alcohol. J. Chem. Soc., 1946, pp. 157-161.
- 26. Dostrovsky, I.; and Hughes, E. D.: Mechanism of Substitution at a Saturated Carbon Atom. Part XXVII. The Role of Steric Hindrance. Sec. B. A Comparison of the Rates of Reaction of Ethyl and neoPentyl Bromides with Inorganic Iodides in Acetone Solution. J. Chem. Soc., 1946, pp. 161-164.
- Dostrovsky, I.; and Hughes, E. D.: Mechanism of Substitution at a Saturated Carbon Atom. Part XXVIII. The Role of Steric Hindrance. Sec. C. A Comparison of the Rates of Reaction of Methyl, Ethyl, n-Propyl, isoButyl, and neoPentyl Bromides with Aqueous Ethyl Alcohol. J. Chem. Soc., 1946, pp. 164-165.
- Dostrovsky, I.; and Hughes, E. D.: Mechanism of Substitution at a Saturated Carbon Atom. Part XXIX. The Role of Steric Hindrance. Sec. D. The Mechanism of the Reaction of neoPentyl Bromide with Aqueous Ethyl Alcohol. J. Chem. Soc., 1946, pp. 166-169.
- 29. Dostrovsky, I.; and Hughes, E. D.: Mechanism of Substitution at a Saturated Carbon Atom. Part XXXI. The Role of Steric Hindrance. Sec. F. A Comparison of the Rates of Reaction of Methyl, Ethyl, n-Propyl, and neoPentyl Bromides with Wet Formic Acid. J. Chem. Soc., 1946, pp. 171-173.

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Test fluid	38 ⁰ C	38 [°] C 99 [°] C		Thermal	
	Kinematıc m ² /sec	viscosıty, e (cS)	g/cm ³ decomposit temperatur T _D , ^O C (a)		
3-Tri-n-butylstannyl (diphenyl)	9.98×10 ⁻⁶ (9.98)	2.4×10 ⁻⁶ (2.4)		236	
3-Tri- <u>n</u> -butylsilyl (diphenyl)	22.9×10 ⁻⁶ (22.9)	3.6×10 ⁻⁶ (3.6)		295	
C-ether	25×10 ⁻⁶ (25)	4.1×10 ⁻⁶ (4.1)	1.18	390	
Five-ring polyphenyl ether	360×10 ⁻⁶ (360)	13×10 ⁻⁶ (13)	1.19	443	

TABLE I. - TYPICAL PROPERTIES OF EXPERIMENTAL FLUIDS

^aMaximum T_D obtained with 3-tri-n-octyl silyl (diphenyl), 358^o C; maximum T_D obtained with 4-triphenyl stannyl (diphenyl), 297^o C.

TABLE II. - THERMAL DECOMPOSITION TEMPERATURES (FROM ISOTENISCOPE) FOR

VARIOUS ORGANOTIN AND ORGANOSILICON COMPOUNDS

Type of	Substituent ^a								
compound	Methyl	Ethyl	<u>n</u> -propyl	<u>n</u> -butyl	<u>n</u> -hexyl	<u>n</u> -octyl	2-Ethyl hexyl	Phenyl	Neopentyl
	т	herma	l decompo	sition ter	nperatur	e (from i	sotenisco	ope), T _D	,, °C
Meta-trialkylstannyl	^b 230		^{b, c} 236	236	^b 236		^b 260		^b 283
Para-trialkylstannyl			^b 240	^b 240				^b 297	
Meta-trialkylsilyl	278	^b 296	276	295	^ь зоо	358			
Para-trialkylsilyl	^b 311			^b 320					

^aSubstituents given in order of increasing steric hindrance.

^bValue corrected for dilution effects (compound was mixed with a four-ring polyphenyl ether to produce a 20 to 50 vol. % concentration).

^CValue for 3, 3' disubstituted derivative was 214[°] C.

TABLE III. - RELATIVE REACTION RATES

FOR BASIC ETHANOLYSIS OF ALKYL

BROMIDES (R-Br) AS FUNCTION OF

SIZE OF ALKYL GROUP (R)

[Data from refs. 25 to 29.]

Alkyl group	Reaction rate relative to that of methyl group
Methyl	1.0
Ethyl	5. 7×10^{-2}
n-propyl	1. 6×10^{-2}
Isobutyl	1. 7×10^{-3}
Neopentyl	2. 4×10^{-7}



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Figure 2. - Friction and wear apparatus.







Figure 4. - Sample cell assembly.



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Figure 7. - Coefficient of friction and rider (ball) wear rate for 3-tri-n-butylstannyl (diphenyl), 3-tri-n-butylsilyl (diphenyl), a polyphenyl ether, and a C-ether Load, 1 kilogram; sliding speed, 17 meters per minute (100 rpm); test duration, 25 minutes; M-50 steel specimens, dry-air atmosphere



Figure 8. - Rider (ball) wear rates for four fluids at four disk temperatures in dry air Load, 1 kilogram; sliding speed, 17 meters per minute (100 rpm); test duration, 25 minutes; M-50 steel specimens

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