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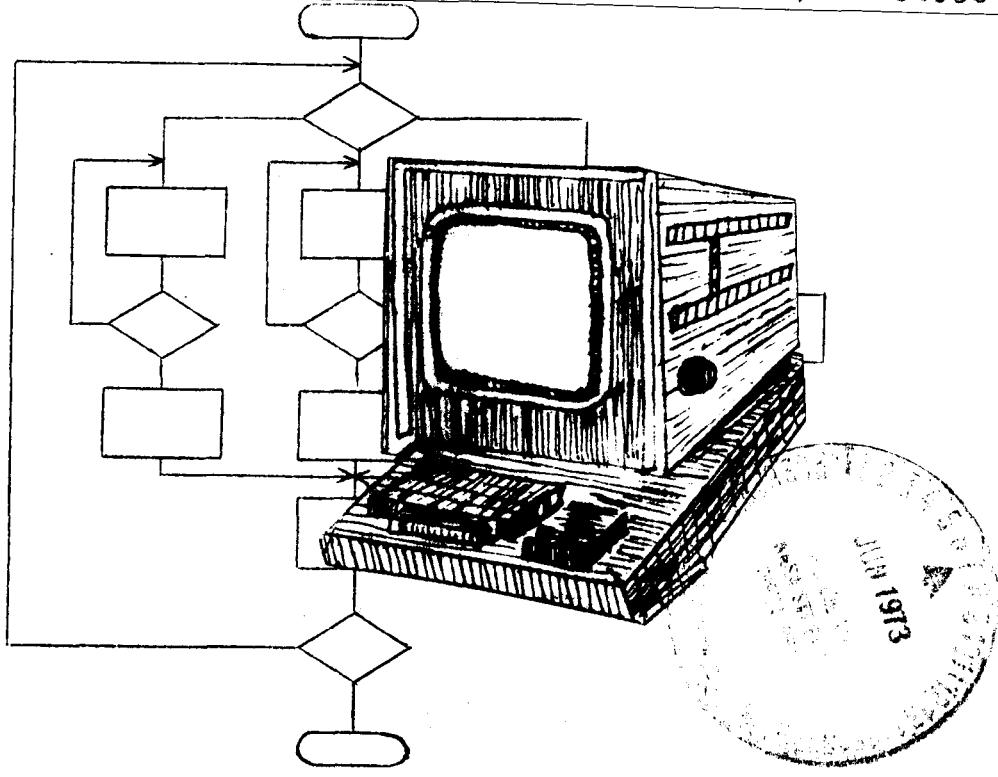
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**NASA**

# Automatic System for Computer Program Documentation

VOLUME 2

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DYNASOR II - A FINITE ELEMENT PROGRAM FOR THE  
DYNAMIC NONLINEAR ANALYSIS OF SHELLS OF REVOLUTION

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USER MANUAL

October 15, 1970

TEES-RPT-70-19

Texas A&M University  
College Station, Texas

II

APPENDIX J

SAMPLE DOCUMENTED PROGRAM

I

## ABSTRACT

The DYNASOR II program is used for the Dynamic Nonlinear Analysis of Shells Of Revolution. The equations of motion of the shell are solved using Houbolt's numerical procedure. The displacements and stress resultants can be determined for both symmetrical and asymmetrical loading conditions. Asymmetrical dynamic buckling can be investigated. Solutions can be obtained for highly nonlinear problems utilizing as many as five of the harmonics generated by SAMMSOR program. A restart capability allows the user to restart the program at a specified time.



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## System Overview

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**SOR - Shell Of Revolution****Computer Programs**

A family of compatible computer codes for the analysis of the shell of revolution (SOR) structures has been developed by researchers at Texas A&M University. These analyses employ the matrix displacement method of structural analysis utilizing a curved shell element. Geometrically nonlinear static and dynamic analyses can be conducted using these codes. The important natural frequencies and mode shapes can also be determined by employing another of the codes. Efficient programming provides codes capable of performing these desired analyses in relatively small amounts of computer time.

Each of these programs has been extensively tested using problems the solutions to which have been reported by other researchers in order to establish the validity of the codes. In addition, the capabilities of the codes have been demonstrated in a number of publications by presenting solutions to problems which were unsolved by other researchers.

SAMMSOR II - Stiffness And Mass Matrices for Shells Of Revolution are generated utilizing the first member of this family. This program accepts a description of the structure in terms of the coordinates and slopes of the nodes and the properties of the elements joining the nodes. For shells with simple geometries (such as cylinders, shallow caps, hemispheres, etc.) the shell geometry can be internally generated. Utilizing the element properties, the structural stiffness and mass matrices are generated for as many as twenty harmonics and stored on magnetic tape. Each of the other SOR programs utilizes the output tape generated by SAMMSOR as input data for the respective analyses. One advantage of creating the stiffness and mass matrices in a separate program is that a variety of analyses can be performed on the same shell configuration without having to create the matrices more than once. Obviously, a variety of boundary and loading conditions can be employed without having to create new mass and stiffness matrices for each case.

SNASOR II - The Static Nonlinear analysis of Shells Of Revolution subjected to arbitrary mechanical and thermal loading is performed using the second computer code. Utilizing the stiffness matrices generated by SAMMSOR and the loading conditions and boundary conditions input to SNASOR II, the equilibrium equations for the structure are generated. The nonlinear strain energy terms result in pseudo generalized forces (as functions of the displacements) which are combined with the applied generalized forces. The resulting set of nonlinear algebraic equilibrium equations is solved by one of several methods: Newton-Raphson

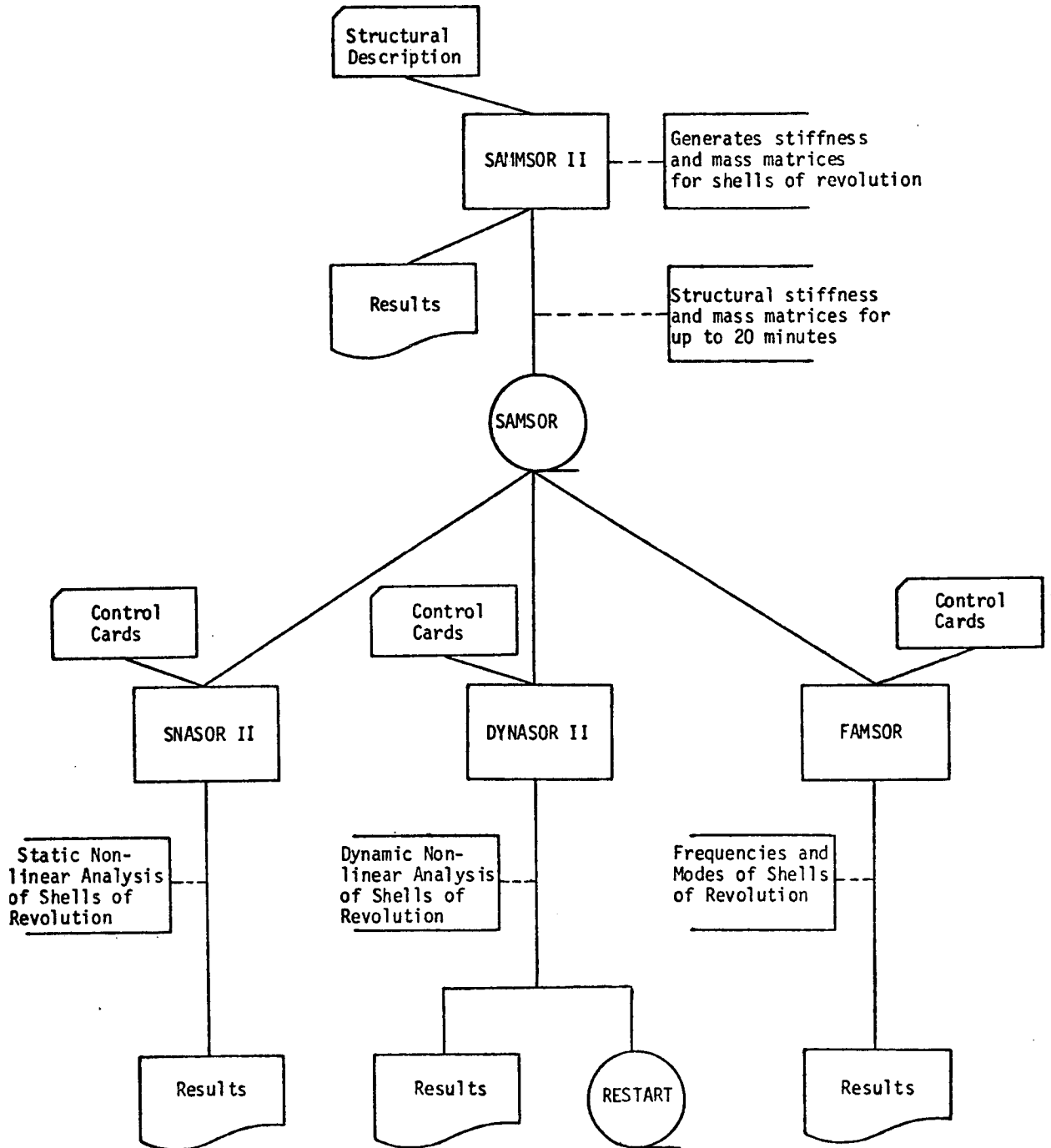
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type iteration, incremental stiffness method, or a modified incremental stiffness method. In general, the Newton-Raphson procedure is the best and yields accurate results for highly nonlinear problems.

DYNASOR II - The third code is used for the Dynamic Nonlinear Analysis of Shells Of Revolution. The equations of motion of the shell are solved using Houbolt's numerical procedure with the nonlinear terms being moved to the right-hand side of the equilibrium equations and again treated as generalized loads. The displacements and stress resultants can be determined for both symmetrical and asymmetrical loading conditions. Asymmetrical dynamic buckling can be investigated using this program. Solutions can be obtained for highly nonlinear problems in reasonable periods of time on the computer utilizing as many as five of the harmonics generated in SAMMSOR. A restart capability is incorporated in this code which allows the user to restart the program at a specified time without having to expend the computer time necessary to regenerate the prior response.

FAMSOR - Frequencies And Modes for Shells Of Revolution can be determined using the fourth code. Using the stiffness matrix generated by SAMMSOR and a lumped mass representation developed from the consistent mass matrix generated by SAMMSOR, a specified number of natural frequencies (beginning with the lowest or fundamental frequency) are obtained using the inverse iteration method. The mode shapes for each of the frequencies are also obtained.

SYSTEM FLOWCHART



## ENVIRONMENT

The DYNASOR II program runs under OS/360 MFT or MVT and requires 220K of memory on an IBM S/360 computer. The system must also have a card reader, printer, 3 9-track tape drives and 2314 disk storage.



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## INTRODUCTION

The DYNASOR II (Dynamic Nonlinear Analysis of Shells Of Revolution) code has been developed to determine the time varying response of shells of revolution to a variety of loading conditions. The code utilizes the stiffness and mass matrices created by the SAMMSOR code for selected harmonics, generates generalized forces from a mechanical and thermal load history, and solves the resulting initial value problem. This report is a user's guide for the DYNASOR II code and is divided into four self-contained sections with an extended appendix.

The first section describes the method of analysis used to obtain the displacements, stresses, and stress resultants for the desired time increments. The formulation of the equations of motion is presented along with the numerical technique employed to obtain the solution these equations.

A section is then presented to enumerate the limitations of the code and to provide valuable guidelines to aid the user in performing the desired analyses. The limitations result partly from the procedures utilized in the method of analysis and partly from the storage capacity and programming procedures employed.

A description of the input data required by the DYNASOR II code is presented in the third section. Examples are provided in instances where the wording might, at first glance, appear to be unclear or insufficient. The limitations placed upon the input parameters are once again enumerated.

The final section contains selected example problems which are designed to illustrate the wide variety of input variations allowed by the code. A copy of the input data required for each of the cases is presented along with selected values of the output data. A thorough understanding of these example problems is mandatory if the user is to become adept at operating the code.

The extended appendix which follows the main report should prove to be extremely helpful if a thorough understanding of the program is desired. A description of the subroutines and the significant Fortran variables is supported by the presence of the subroutine call map and a flow chart of the basic operations of the code. The sections describing the restart capability and the specification of the loads should prove invaluable to users who desire to obtain optimum performance from the code. A discussion of the program output is then followed by a description of the changes necessary to modify the capacity of the code.

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## SECTION I

## METHOD OF ANALYSIS

## Introduction

The purpose of this section is to provide theoretical documentation of the equations and procedures employed in the DYNASOR II code to perform the Dynamic Nonlinear Analysis of Shells of Revolution. The matrix displacement method of a structural analysis is utilized. Since the documentation for the development of the stiffness and mass matrices has been adequately presented in the SAMSOR II user's manual,<sup>1</sup> this section will not attempt to duplicate the previous presentation. The dynamic equations of motion are derived and the numerical techniques utilized to effect the solution of these equations are discussed.

## Equations of Motion

The matrix displacement method is an energy formulation and, consequently, the equations of equilibrium for the nonlinear dynamic response are obtained from Lagrange's equation:

$$\frac{d}{dt} \left( \frac{\partial T}{\partial \dot{q}_i^n} \right) + \frac{\partial U}{\partial q_i^n} = Q_i^n \quad (1)$$

where

$q_i^n$  = generalized degree of freedom  $i$  of harmonic  $n$

$T$  = kinetic energy

$U$  = internal energy (2)

$Q_i^n$  = generalized force for degree of freedom  $i$  of harmonic  $n$

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Since the internal energy of a structure is a scalar quantity, the expression for this quantity may be separated into various parts. The formulation used in this analysis considers the internal energy as

$$U = U_L + U_{NL} - (U_L^t + U_{NL}^t) \quad (3)$$

where the superscript, t, denotes the inclusion of thermal effects and

$U_L$  = strain energy based upon linear strain displacement relations

$U_{NL}$  = strain energy due to the inclusion of nonlinear contributions in the strain displacement relations

By substituting Eq. 3 into Eq. 1 and taking the nonlinear strain energy terms to the right-hand side, the equations of motion for the nonlinear dynamic analysis of shells of revolution can be written in matrix form as

$$[M^n]\{q^n\} + [K^n]\{q^n\} = \{Q^n\} + \{Q_t^n\} - \left\{ \frac{\partial U_{NL}}{\partial q^n} \right\} + \left\{ \frac{\partial U_{NL}^t}{\partial q^n} \right\} \quad (4)$$

The column matrix,  $\{Q_t^n\}$ , of pseudo linear thermal loads is evaluated exactly from  $\{\partial U_L^t / \partial q^n\}$ . It should be noted that Eq. 4 is valid for any harmonic n with the coupling between the harmonics appearing on the right-hand side. In this formulation the nonlinear terms are treated as pseudo generalized forces which are applied to the structure. The obvious advantage of this formulation is that a tremendous savings in computer time can be realized since the stiffness matrix does not change as the displacements vary and must, therefore, be calculated only once. With most other formulations for geometric nonlinearities, the stiffness matrix must be updated at each time step.

Strain Displacement Relations

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The DYNASOR II code utilizes the strain displacement equations given by Novozhilov<sup>2</sup> as restricted to shells of revolution with the additional assumption being made that the only important nonlinear contributions arise from rotations about the shell coordinate axes. The midsurface strain expressions can then be written as

$$\begin{aligned}\epsilon_s &= \hat{e}_s + \frac{1}{2} \hat{e}_{13}^2 \\ \epsilon_\theta &= \hat{e}_\theta + \frac{1}{2} \hat{e}_{23}^2 \\ \epsilon_{s\theta} &= \hat{e}_{s\theta} + \hat{e}_{13} \hat{e}_{23}\end{aligned}\tag{5}$$

The changes in curvature are those used in linear theory

$$\begin{aligned}\chi_s &= -\partial \hat{e}_{13} / \partial s \\ \chi_\theta &= -(1/r)(\partial \hat{e}_{23} / \partial \theta) - (1/r) \sin \phi \hat{e}_{13} \\ \chi_{s\theta} &= -(1/r)(\partial \hat{e}_{13} / \partial \theta) + (\sin \phi / r) \hat{e}_{23} - \partial \hat{e}_{23} / \partial s\end{aligned}\tag{7}$$

#### Pseudo Nonlinear Forces

The nonlinear terms in this analysis are treated in the same way as the generalized forces due to external loading. The generalized forces due to the nonlinearities are evaluated for each element and are then combined at the nodes. A detailed presentation of the procedures utilized in calculating the nonlinear forces has been made in Ref. 3 with an overview of the same material being provided in Ref. 4.

The pseudo forces are obtained by retaining strain energy terms containing the rotations raised to the fourth power. The retention of the fourth order terms has been shown<sup>5</sup> to be absolutely essential in cases where the nonlinear terms are substantial. The results presented in Ref. 6 for static shell analysis did not include the effects of the fourth order terms but results obtained after the incorporation of these terms revealed once again the necessity of retaining these contributions.

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The generalized forces due to nonlinearities are evaluated using linear displacement functions in the variables  $u$ ,  $v$ , and  $w$  and employing strip integration over the length of the element. The integrals around the circumference are evaluated in closed form for the particular harmonics chosen. This procedure is simpler than the one employed in Ref. 6 and permits the nonlinear forces to be evaluated without the use of secondary storage on the computer. Detailed justification for this simplified procedure has been made in Ref. 3 so these arguments will not be enumerated again. It will suffice to note that due to the exact evaluation of the integrals in the circumferential direction, it is reasonable to expect rapid convergence as the number of harmonics is increased. Examples have shown (Ref. 3) that the use of the strip integration over the length of the element produces convergence quite rapidly as the number of elements is increased.

#### Thermal Terms

The temperature distribution and the temperature gradients in the normal direction for an element are expanded in a Fourier series in a manner similar to that used for the displacement functions. The temperatures and temperature gradients for an element are assumed constant over each element in the meridional direction with step variations allowed in the circumferential direction. In cases where the step variation in the circumferential direction is not considered accurate enough, the Fourier coefficients may be specified as input information.

The linear and nonlinear contributions are separated with the linear thermal loads for each harmonic being evaluated as

$$\{Q_t^n\} = \left\{ \frac{\partial U_L^t}{\partial q^n} \right\} \quad (8)$$

Employing a coordinate transformation to change to partial derivatives with respect to the generalized shell coordinates, the problem reduces to the evaluation of the partial derivatives of  $U_L^t$  with respect to the coefficients  $\alpha_1, \alpha_2, \dots, \alpha_8$ . These partial derivatives are presented in Eqs. 26 of Ref. 3, and the terms of  $\{Q_t^n\}$  are listed in the appendix of the same report.

The nonlinear thermal loads are treated in essentially the same manner as the generalized forces due to nonlinearities are treated. Utilizing the same approximations as for the nonlinearities due to applied forces, the expression for the nonlinear thermal contribution is given by Eq. 28 of Ref. 3.

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## Stress Resultants

In this code, the stress resultants are determined by the use of the assumed displacement functions and finite difference relations at the mid-point of each element.

For orthotropic shells the stress resultants may be written as

$$\begin{Bmatrix} N_s \\ N_\theta \\ N_{s\theta} \\ M_s \\ M_\theta \\ M_{s\theta} \end{Bmatrix} = \begin{bmatrix} C_1 & \nu_{s\theta} C_1 & 0 & 0 & 0 & 0 \\ \nu_{\theta s} C_2 & C_2 & 0 & 0 & 0 & 0 \\ 0 & 0 & G_1 & 0 & 0 & 0 \\ 0 & 0 & 0 & D_1 & \nu_{s\theta} D_1 & 0 \\ 0 & 0 & 0 & \nu_{\theta s} D_2 & D_2 & 0 \\ 0 & 0 & 0 & 0 & 0 & G_2 \end{bmatrix} \begin{Bmatrix} \epsilon_s \\ \epsilon_\theta \\ \epsilon_{s\theta} \\ \chi_s \\ \chi_\theta \\ \chi_{s\theta} \end{Bmatrix}$$

where

$$\begin{aligned} C_1 &= E_s t / (1 - \nu_{s\theta} \nu_{\theta s}) & C_2 &= E_\theta t / (1 - \nu_{s\theta} \nu_{\theta s}) \\ G_1 &= Gt & G_2 &= Gt^3 / 12 \\ D_1 &= E_s t^3 / [12(1 - \nu_{s\theta} \nu_{\theta s})] & D_2 &= E_\theta t^3 / [12(1 - \nu_{s\theta} \nu_{\theta s})] \end{aligned} \quad (10)$$

The shear resultants are determined approximately from the equations

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of the undeformed shell as

$$\begin{aligned}
 Q_s &= \frac{1}{r} \left[ \frac{\partial}{\partial s} (rM_s) + \frac{\partial M_{s\theta}}{\partial s} - M_{\theta} \sin\phi \right] \\
 Q_{\theta} &= \frac{1}{r} \left[ \frac{\partial}{\partial s} (rM_s) + \frac{\partial M_{\theta}}{\partial \theta} + M_{s\theta} \sin\phi \right]
 \end{aligned}
 \tag{11}$$

#### Numerical Solution of Equations of Motion

Since a closed-form solution of Eq. 4 is generally not available, a numerical method must be used to determine the solution to the equations of motion. A finite difference procedure developed by Houbolt (Ref. 7) has been selected for use in the DYNASOR II code.

The equations of motion, Eq. 4, can be reduced to a system of equations of the form

$$[M]\{\ddot{q}\} + [K]\{q\} = \{F(t,q)\} \tag{12}$$

The load matrix  $\{F(t,q)\}$  is equivalent to the right-hand side of Eq. 4. The initial displacements and velocities of the nodes must be specified and can be written as

$$q_0 = \{q\}_0 \tag{13}$$

$$\dot{q}_0 = \{\dot{q}\}_0$$

Utilizing the Houbolt procedure, the accelerations of the nodes of the shell are approximated by a third-order backwards difference

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expression

$$\{\ddot{q}_{n+1}\} = \frac{1}{(\Delta t)^2} \{2q_{n+1} - 5q_n + 4q_{n-1} - q_{n-2}\} \quad (14)$$

Substitution of Eq. 14 into Eq. 12 yields the following expression which is utilized to solve for the displacements at the end of each time step, except the first one:

$$\begin{aligned} (2[M] + (\Delta t)^2[K])\{q_{n+1}\} &= (\Delta t)^2\{F(t,q)_{n+1}\} \\ &+ [M]\{5q_n - 4q_{n-1} + q_{n-2}\} \end{aligned} \quad (15)$$

To determine the displacements at the end of the first time step, the following equation is employed

$$\begin{aligned} (6[M] + (\Delta t)^2[K])\{q_1\} &= (\Delta t)^2\{F(0,q_0)\} \\ &+ [M]\{2(\Delta t)^2\ddot{q}_0 + 6\Delta t\dot{q}_0 + 6q_0\} \end{aligned} \quad (16)$$

It should be noted that the selection of the Houbolt procedure for inclusion in the code was made only after evaluating the advantages and disadvantages of a number of solution schemes (Ref. 9). The Houbolt procedure proved to be the only method capable of providing stable solutions for highly nonlinear problems while utilizing a reasonably large time increment. The significant observations made in Ref. 8 concerning Houbolt's procedure will now be presented.

It was found that double precision arithmetic is necessary if the code is utilized for highly nonlinear problems on an IBM 360/65 system (or comparable system). It is believed that if the DYNASOR II code is used on computers which have a longer word length than the 360/65 system (such as the CDC 6600) double precision arithmetic will not be necessary. Utilizing the Houbolt scheme, it has been shown that the solution converges as the number of elements is increased. Although the Houbolt procedure has been shown to be unconditionally stable for the linear problem, it has found that this is not the case with the nonlinear formulation. The damping inherent in the Houbolt procedure was noted in some instances, but the savings in computer time resulting



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from employing this procedure far outweighs this slight drawback. Solutions (without the damping) were obtained, in some instances, in one-eighth (1/8) the amount of time required by other procedures. In all cases which were run, stable, undamped solutions were obtained using larger time increments than could be used with the other methods.

### Extrapolation of Forces

In order to employ Eq. 15, the loads at the end of the (n+1)th time step must be known. These loads, because of the presence of the nonlinear terms, are a function of the displacements to be calculated and therefore cannot be evaluated exactly. The right-hand side of Eq. 12 is, therefore, evaluated using a first-order Taylor's series expanded about the n-th increment:

$$\{F(t,q)_{n+1}\} = \{F(t,q)_n\} + \frac{\partial}{\partial t} \{F(t,q)_n\} \Delta t + O(\Delta t)^2 \quad (17)$$

A second-order extrapolation process has been employed (Ref. 8), but the results indicated that the linear extrapolation procedure was more stable.

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## SECTION II

## USER GUIDELINES AND PROGRAM LIMITATIONS

Guidelines for the use of the DYNASOR II code along with the limitations placed upon the analysis are enumerated in this section. Some of these limitations are the result of the procedures used to program the equations while other limitations are inherent in the formulation of the equations. Since most of the limitations are minor in nature, the DYNASOR II code may be used to solve a wide variety of important shell dynamics problems.

The maximum number of elements which the program may use is fifty (50). The maximum number of harmonics which may be coupled for the analysis is five (5). It is believed that these limitations will not hinder the user in solving most problems. However, since undoubtedly some users will want to modify the program capacity, instructions for increasing or decreasing the allowable number of elements and/or harmonics are provided in appendix 8.

In all analyses using the DYNASOR II code, the zeroth (0) harmonic must be specified as one of the input harmonics.

The coefficients of thermal expansion are assumed to be constant in the two principal directions for any given element but may vary from element to element.

The number of nodal restraints must be less than or equal to the maximum number of degrees of freedom for each harmonic (204).

The displacements of the nodes may be calculated for as many as twenty (20) angles around the circumference of the shell element.

While the displacements are calculated at every time increment, it is necessary to calculate the stresses only at time steps where a printout of the stresses is desired. The stresses and stress resultants are calculated at the middle of the elements (s-direction) for up to twenty (20) angles in the circumferential direction. The angles at which the stresses are calculated are the same as those at which the displacements are determined. The stresses on both the inner and outer surfaces are determined.

The units used in the program must be consistent with those used in the SAMNSOR code. All calculations in the versions supplied to the users of the code are given inch-pounds-seconds units.

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The program accepts the mechanical and thermal load histories by accepting descriptions at discrete points in time. The difference between the times for which loads are specified must, in all cases, be greater than the value of the time increment used in solving the equations of motion. The load variation curve is approximated by assuming a linear variation of the generalized forces between the times at which the loads are specified. It may therefore be necessary to specify the loads and temperatures at a fairly large number of points in time if the loads vary rapidly with time.

If the loads and/or temperatures propagate in any direction (moving loads), it will also be necessary to specify the loads at a fairly large number of points in time.

Pressure loadings, temperatures, and temperature gradients are assumed to be constant over the meridional length of the element but may vary in the circumferential direction. The variation in the circumferential direction (except for shear loadings) must be symmetric about the meridian corresponding to  $\theta = 0$  degrees. These loadings may be input either by specifying the values at a number of circumferential angles for each element or by specifying the values of the Fourier coefficients for each harmonic.

If the program is not being restarted, the loads and temperatures must be specified at time  $T1 = 0.0$ . Times at which loads must be specified when restarting the program are noted in Appendix 8.

One of the most important considerations in any dynamic analysis is the selection of the time increment to be used in the analysis. Several criteria have been developed for use in selecting a time increment in analyses utilizing finite difference techniques. Most of these criteria require that the time increment be less than the time required for a signal to travel at the speed of sound from one difference point to the next. These criteria have been found (ref. 8) inadequate for use in this analysis. A "feel" for the selection of a time increment must be obtained by the user. To facilitate the development of this "feel" the time increments utilized in a number of problems have been carefully documented in Refs. 3 and 4. In addition, the input data for the example problems should prove helpful.

A restart capability is incorporated in the code to enable the user to calculate the response from a specified point in time without having to recalculate the response prior to this time. A most valuable use of this capability arises if, after evaluation of the results of a run, it is decided to extend the calculations to observe more cycles of response. If it is desired to employ a different time increment (either smaller or larger), the user should refer to the discussion in Appendix 5. Effective use of the restart capability can result in a substantial savings of computer time. In general, the information necessary for restarting the code should be placed on tape at least every 100-400 time

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increments to insure that the information will be available if it is deemed desirable to restart the program.

The pseudo loads due to the nonlinearities associated with the initial displacements are neglected when calculating the response at the end of the first time step. However, when restarting the code, the initial increment utilizes both the mechanical and pseudo forces.

An extended effort has been made to check all aspects of the code. Comparisons of the response obtained using DYNASOR II with the results obtained by other researchers are presented in Ref. 3 and 4. These comparisons firmly establish the validity of the code. Although the programming logic and the formulation have been thoroughly checked to insure the correctness of the code, the authors assume no responsibility for the results obtained using the code.

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## SECTION III

## PROGRAM INPUT

The DYNASOR II code has been written so that the code can be employed by researchers who are not familiar with the inner workings of the program. Utilizing the guidelines and adhering to the limitations presented in the previous section, it is believed that most users will find it relatively easy to employ the code.

The code is available in the FORTRAN IV language using double precision or single precision arithmetic. This double precision version requires a storage space of about 330K bytes on IBM 360/65 system while the single precision storage space is about 200K bytes. Efforts have been made to make this code compatible with a large number of computing systems. In particular, adaption of the code for use on a CDC 6600 computer requires only minor changes.

The input data for a run consists of one card I (card types will be explained on the following pages) followed by a complete set of data (cards II-X) for each case. The set of cards II-X is the input data required to generate the response of a shell for a given number of harmonics due to a particular loading. The cards comprising the data deck for both an initial run and a restart are schematically represented in Fig. 1. The cards specifying the Fourier harmonics, the initial conditions, and the boundary conditions are omitted from the input deck when using the restart mode. If more than one case is to be run, include a set of data for each of the cases. There is no limit on the number of cases which may be included in a run. A card must be placed at the end of the data for the final case.

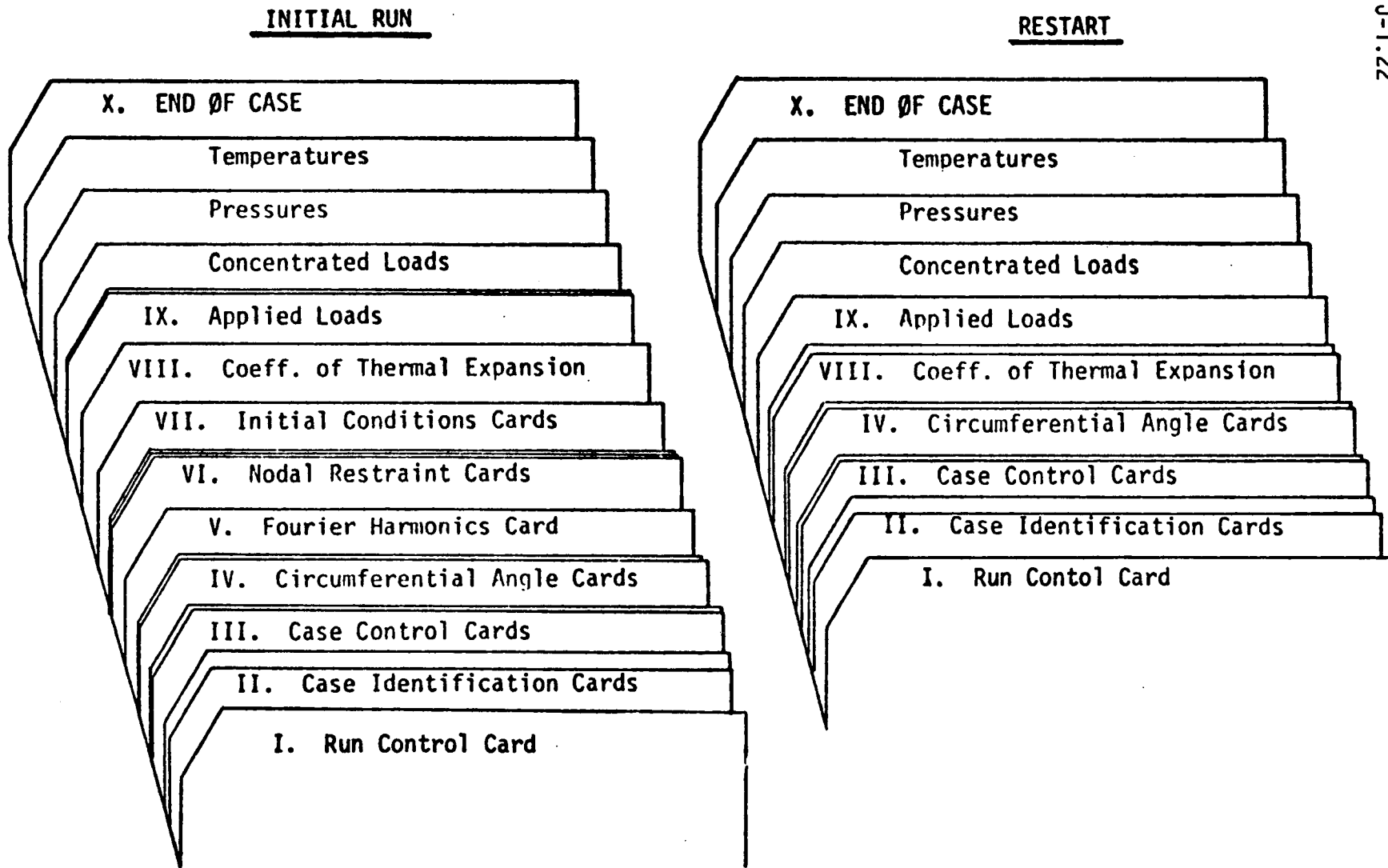


FIG. 1 CONSTITUTION OF DATA DECKS - INITIAL RUN AND RESTART MODES.

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## I. RUN CONTROL CARD

This card is used to identify the number of cases to be run and the logical unit numbers of the scratch tapes used in the run. (ONLY ONE CARD I IS USED PER RUN.)

Card Type I Format (3I5)		
Columns	Variable	Description
1-5	NCASES	The number of different data sets utilized for this run. .....
6-10	ND	Logical unit number of the scratch tape onto which all the data is read at the start of the run. .....
11-15	NS	Logical unit number of a second scratch tape used by the program. .....

## II. CASE IDENTIFICATION CARDS

These cards allow the user to print out comments which identify the problem being run.

A. Control Card (ONE CARD II-A PER DATA SET)

Card Type II-A Format (2I5)		
Columns	Variable	Description
1-5	NCARDS	Number of comment cards (TYPE II-B) which follow. .....
6-10	NT	Logical unit number of the tape (prepared by SAMMSOR) from which the stiffness and mass matrices, element properties, and re-start information, if needed, will be read. .....

B. Identification Cards - The information punched on these cards is printed as output and should identify the problem being run. These comments should not duplicate those of the SAMMSOR case since the SAMMSOR comments will also appear as output. (IF NCARDS=0, OMIT CARDS II-B, OTHERWISE INCLUDE NCARDS OF TYPE II-B.)

Card Type II-B Format (20A4)		
Columns	Variable	Description
1-80	COMENT	Any desired alphanumeric information may be printed on these cards. .....



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III. CASE CONTROL CARDS

A. Control Constants - Time parameters, restart information, and other miscellaneous control constants are input on this card. (INCLUDE ONE CARD III-A PER DATA SET.)

Card Type III-A Format (2F10.0,4I5)		
Columns	Variable	Description
1-10	TOTIME	The maximum time (seconds) for which the calculations are to be performed. .....
11-20	DELTE	Time increment (seconds) used in solving the equations of motion. .....
21-25	IRSTRT	Control constant which indicates if the solution is being restarted. If the solution is being restarted set IRSTRT = 1. If not, set IRSTRT = 0. .....
26-30	INCRST	The number of the time increment at which the solution is to be restarted. INCRST must be an integer multiple of the value of NPRNIT used in the previous run. If IRSTRT = 0, set INCRST = 0. .....
31-35	NCLOSE	For a closed shell (such as a spherical cap or a hemisphere) where node 1 is at the apex, set NCLOSE = 1. Radial and rotational restraints will then be applied for the zeroth harmonic to aid the numerical stability of the solution. If the shell does not fit the above description, set NCLOSE = 0. .....
36-40	ITELF	If thermal loads are to be applied in the program, set ITEL = 1. Otherwise, set ITEL = 0. .....

B. Print Control Card - The constants used to control the program output are punched on this card. (INCLUDE ONE CARD III-B PER DATA

SET.)

Card Type III-B Format (10I5)		
Columns	Variable	Description
1-5	NPRNTQ	If the displacements are to be printed, set NPRNTQ = 1. If not, set NPRNTQ = 0. .....
6-10	IPRINT	If NPRNTQ = 1, the displacements will be printed every IPRINT time increments beginning with the first time step. If NPRNTQ = 0, set IPRINT = 0. .....
11-15	NCLCST	If the stresses and stress resultants are to be calculated, set NCLCST = 1. If not, set NCLCST = 0. .....
16-20	NSTRSS	If NCLCST = 1, the stress and stress resultants will be calculated and printed every NSTRSS time increments beginning with the first step. If NCLCST = 0, set NSTRSS = 0. .....
21-25	NPRNT	If restart information is to be placed on tape, set NPRNT = 1. If not, set NPRNT = 0. .....
26-30	NPRNIT	If NPRNT = 1, the restart information will be written on the output tape every NPRNIT time increments. If NPRNT = 0, set NPRNIT = 0. It is suggested that relatively large values of NPRNIT be used, say 200, 400, etc., if the total number of time steps is relatively large. .....
31-35	NPRNTL	If a printout of the applied loads is desired, set NPRNTL = 1. Otherwise, set NPRNTL = 0. .....
36-40	NPRNTF	If a printout of the generalized forces is desired, set NPRNTF = 1. Otherwise, set NPRNTF = 0. .....

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41-45	NPRNTH	If the Fourier coefficients for the temperature and temperature gradient are to be printed, set NPRNTH = 1. Otherwise, set NPRNTH = 0. .....
46-50	NPRNMS	If the mass and stiffness matrices are to be printed, set NPRNMS = 1. If not, set NPRNMS = 0. .....

IV. CIRCUMFERENTIAL ANGLE CARDS

The circumferential angles at which the displacements and stresses are to be calculated are read from these cards.

A. Control Card - (ONE CARD IV-A PER DATA SET.)

Card Type IV-A Format (I5)		
Columns	Variable	Description
1-5	NTHETA	The number of circumferential angles at which the displacements and possibly stresses are to be calculated. ( $1 \leq NTHETA \leq 20$ ) .....

B. Circumferential Angles - (INCLUDE 1-3 CARDS IV-B PER DATA SET DEPENDING UPON THE VALUE OF NTHETA.)

Card Type IV-B Format (8F10.0)		
Columns	Variable	Description
1-10	THETA (1)	Circumferential angles at which the displacements and possible stresses will be calculated. .....
11-20	THETA (2)	(If it is desired to calculate the displacements only along the line = 0, then include one card IV-B and set THETA (1) = 0.0)
"	"	
"	"	
"	THETA (NTHETA)	.....

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V. FOURIER HARMONICS CARD

This card provides the number of Fourier cosine harmonics to be employed for this analysis and enumerates the specific harmonics to be used. (IF IRSTRT = 1, OMIT CARD V. OTHERWISE, INCLUDE ONE CARD V PER DATA SET.)

Card Type V Format (6I5)		
Columns	Variable	Description
1-5	NH	The total number of Fourier cosine harmonics to be utilized in this analysis ( $1 \leq NH \leq 5$ ). .....
6-10	IHARM(1)	Specific harmonics numbers to be employed. NH values must be given and the zero harmonic must always be specified as one of the input harmonic numbers. The user should check to be certain that the information for each of these harmonics has been created and stored on tape by the SAMMSOR code. .....
11-15	IHARM(2)	
16-20	IHARM(3)	
21-25	IHARM(4)	
26-30	IHARM(5)	

Example: Consider a case where it is desired to utilize harmonics 0, 2, 3, and 4. The input data for card V would then utilize the following values:

```

NH           = 4
IHARM(1)    = 0   NOTE:  IHARM(1) should always
                    be set equal to zero.
IHARM(2)    = 2
IHARM(3)    = 3
IHARM(4)    = 4
    
```

Columns 26-30 corresponding to IHARM(5) should be left blank for this example since only four harmonics are being run.

**VI. NODAL RESTRAINT CARDS (Boundary Conditions)**

The displacement constraints applied to the shell are described utilizing these cards. (IF IRSPRT = 1, OMIT CARDS VI-A AND VI-B.)

**A. Control Card - (ONE CARD VI-A PER DATA SET, UNLESS IRSTRT = 1.)**

Card Type VI-A Format (I5)		
Columns	Variable	Description
1-5	NODRES	Total number of displacement constraints to be applied to the shell (0 ≤ NODRES ≤ 204) .....

**B. Boundary Conditions - (THE NUMBER OF CARDS OF TYPE VI-B MUST EQUAL NODRES, UNLESS IRSTRT = 1. IF NODRES = 0, OMIT CARDS VI-B.)**

Card Type VI-B Format (2I5)		
Columns	Variable	Description
1-5	NP	Number of the node where the restraint is to be applied. .....
6-10	NDIRCT	Key used to indicate the degree of freedom which is restrained. NDIRCT = 1 applies axial restraint NDIRCT = 2 applies circumferential restraint NDIRCT = 3 applies radial restraint NDIRCT = 4 applies rotational restraint .....

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VII. INITIAL CONDITIONS CARDS

The initial velocities and displacements of the nodes are specified on these cards. (IF IRSTRT = 1, OMIT CARDS VII-Z, VII-B, AND VII-C.)

- A. Control Card - Utilization of this control card greatly simplifies the specification of the initial conditions if either the initial velocities or the initial displacements, or both, are equal to zero. (ONE CARD VII-A PER DATA SET)

Card Type VII-A Format (2I5)		
Columns	Variable	Description
1-5	IQN	If the initial velocities at all the nodes are zero, set IQN = 0. If not, set IQN = 1. .....
6-10	IQN1	If the initial displacements at all the nodes are zero, set IQN1 = 0. If not, set IQN1 = 1. .....

- B. Initial Velocities - The initial nodal velocities must be specified for each node of the shell for each harmonic to be run. The logic used to input the nodal velocities is essentially the same as the procedure used to specify the element properties in the SAMSOR code. The initial velocities for each of the nodes are specified for the first of the input harmonics, then for the second input harmonic, etc. This process is repeated until the nodal velocities for the first of the input harmonics, then for the second input harmonic, etc. This process is repeated until the nodal velocities for each harmonic have been specified. (IF IQN = 0, OMIT CARDS

VII-B.)

Card Type VII-B Format (2I5, 4F10.0)		
Columns	Variable	Description
1-5	IN1	First node to which the velocities specified on this card are applied. .....
6-10	IN2	Last node to which the velocities specified on this card are applied. .....
11-20	$\dot{q}_1$	Initial nodal velocity in the axial direction for a particular harmonic. .....
21-30	$\dot{q}_2$	Initial nodal velocity in the circumferential direction for a particular harmonic. .....
31-40	$\dot{q}_3$	Initial nodal velocity in the radial direction for a particular harmonic. .....
41-50	$\dot{q}_4$	Initial nodal rotational velocity in the meridional direction for a particular harmonic. .....

C. Initial Displacements - In identically the same manner as is utilized for the initial velocities, the initial displacements are



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specified for each harmonic. (IF IQN1 = 0, OMIT CARDS VII-C)

Card Type VII-C Format (2I5, 4F10.0)		
Columns	Variable	Description
1-5	IN1	First node to which the displacements specified on this card are applied. .....
6-10	IN2	Last node to which the displacements specified on this card are applied. .....
11-20	q 1	Initial nodal displacement in the axial direction for a particular harmonic. .....
21-30	q 2	Initial nodal displacement in the circumferential direction for a particular harmonic. .....
31-40	q 3	Initial nodal displacement in the radial direction for a particular harmonic. .....
41-50	q 4	Initial nodal rotation in the meridional direction for a particular harmonic. .....

VIII. COEFFICIENTS OF THERMAL EXPANSION

If the thermal effects are to be included in the analysis, the coefficients of thermal expansion must be specified using these cards. These coefficients are assumed to be constant for a given element but may vary from element to element. These coefficients are read in the same manner as the element properties in the SAMSOR code. (THE NUMBER OF CARDS VIII MUST BE  $\leq$  NELEMS FOR ANY GIVEN DATA SET. IF ITELP = 0, OMIT CARDS VIII.)

Card Type VIII Format (2I5, 2F10.0)		
Columns	Variable	Description
1-5	IELM1	Number of the first element to which the properties on this card apply. .....
6-10	IELM2	Number of the last element to which the properties on this card apply. .....
11-20	ALSI1	Coefficient of thermal expansion in the meridional direction (in/in/deg). .....
21-30	ALTI1	Coefficient of thermal expansion in the circumferential direction (in/in/deg). .....

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## IX. APPLIED LOADS, TEMPERATURES, AND TEMPERATURE GRADIENTS

Since the concentrated nodal loads, distributed pressures, temperatures, and temperature gradients may vary in time; it may be necessary to specify these loads at a number of points in time. If these loads and temperatures are input at times  $T1_j$  and  $T1_{j+1}$ , the program will calculate generalized forces due to these loads at each of the input times. A linear variation of the generalized forces is then assumed between the times the loads are input. As soon as the value of the time reaches  $T1_{j+1}$ , a new set of loads is read in at  $T1_{j+2}$  and the process of calculating the generalized forces is repeated. The time increment, DELTE (CARD III-A), used in the solution of the equations of motion must be less than the difference between any two of the times at which the loads are specified. If the loads and/or temperatures propagate in and direction (moving loads), it is advisable to specify the loads at more times than is necessary if they vary in intensity only.

Ring loads can be applied at the nodes and must be input for each of the harmonics. The ring loads utilize the same sign convention employed for the shell nodal displacements.

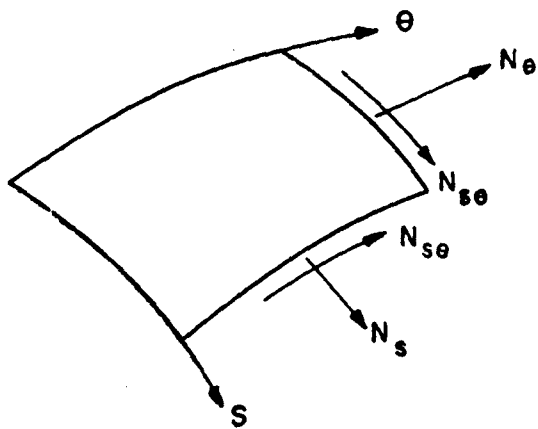
The pressure loadings, temperatures and temperature gradients are assumed constant over the meridional length of the element but variations in the circumferential direction are allowed. These loadings may be input in one of two ways. Either the Fourier coefficients can be specified for each harmonic or the values of the loads may be specified at a number of circumferential angles around the shell elements. Utilizing this second procedure a step function variation is assumed in the circumferential direction. That is, the load is assumed constant from  $\theta_j$  to  $\theta_{j+1}$  with the value of the loads being equal to those specified at  $\theta_j$ . Sign conventions for the pressure loading are given in Figure 2.

A control card (Card Type IX-A) containing several key variables is used to guide the reading of the loading conditions. Proper selection of the values of these key variables results in a highly efficient procedure for specifying a wide variety of loading conditions. The key words and their meanings are explained in Figure 3.

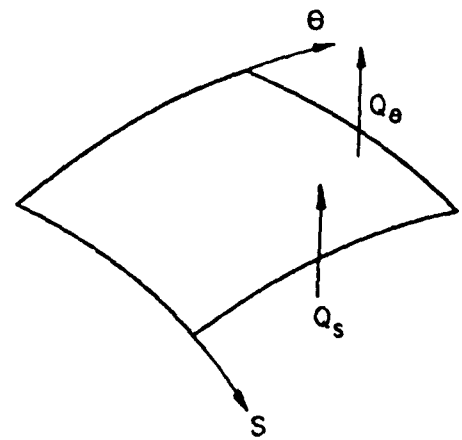
Before attempting to input loads to the code the user is advised to study the guidelines presented in Section II, the example problems of Section II, and Appendix 6 which presents a thorough discussion of the various procedures necessary for specifying the loads.

### A. Load Control Card

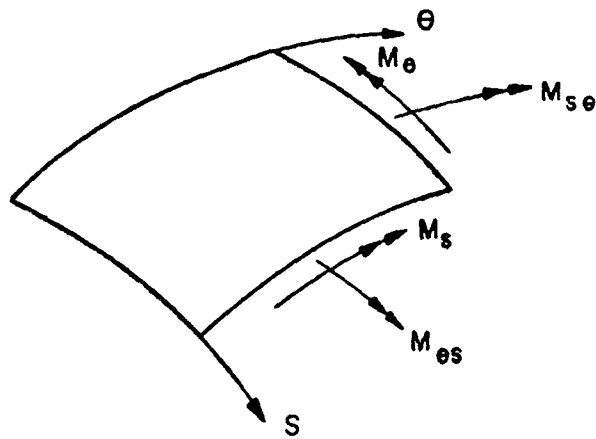
This control card is utilized to direct the input of the loads for a given time. This card indicates the presence or absence of concentrated



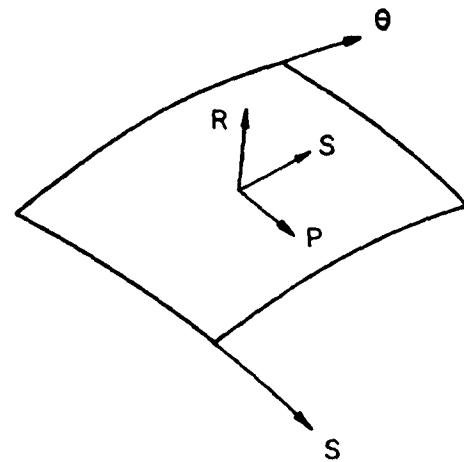
(a) Membrane force resultants



(b) Transverse force resultants



(c) Moment resultants



(d) Loads per unit area

FIG 2 POSITIVE DIRECTION OF FORCES, MOMENTS, AND LOADS ON SHELL SEGMENT

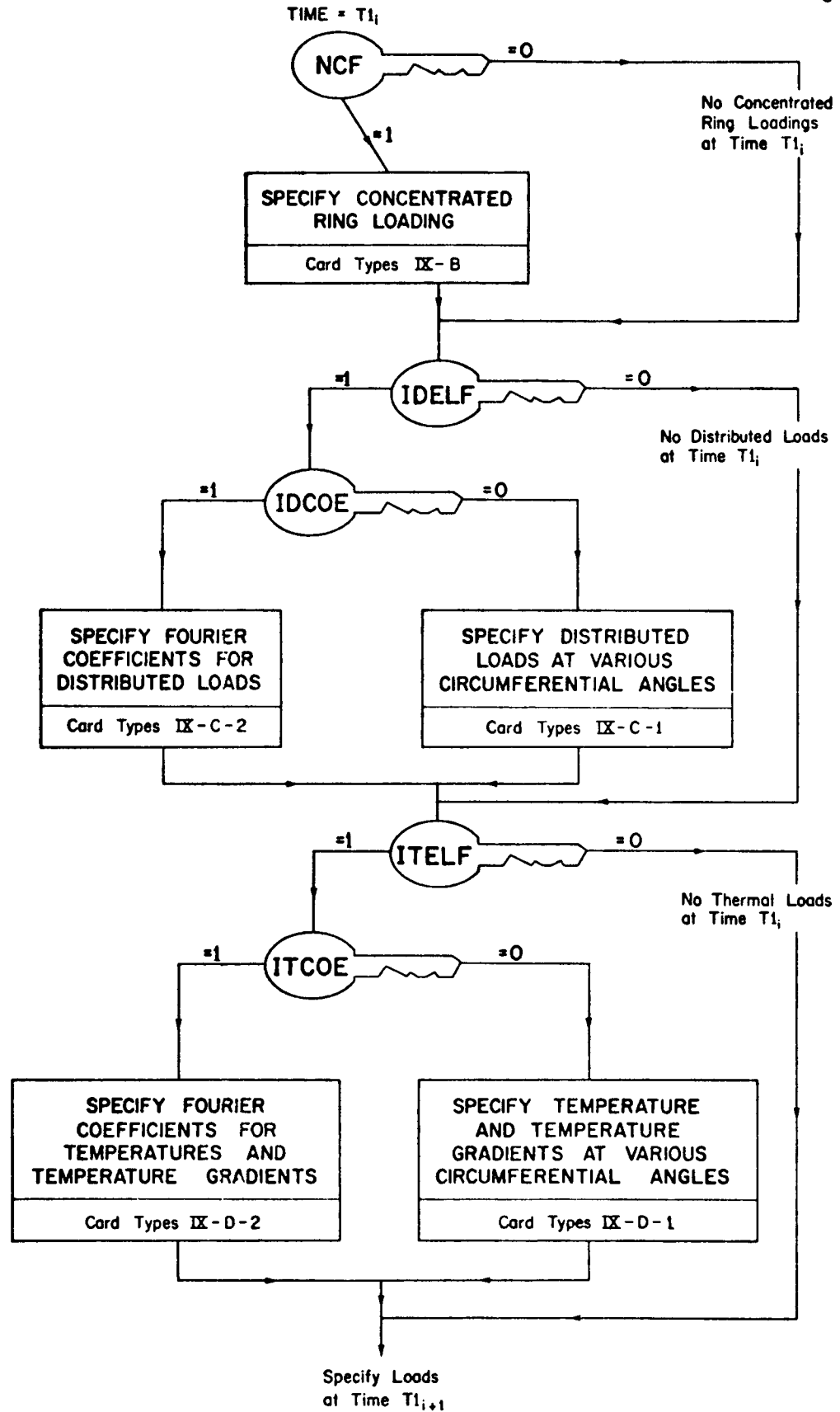


Fig. 3 LOAD SPECIFICATION AT TIME  $T_{1i}$

forces and distributed pressure loadings and indicates the procedure to be utilized for creating the generalized thermal forces. (ONE CARD IX-A IS NECESSARY FOR EACH TIME AT WHICH THE LOADS ARE BEING INPUT.)

Card Type IX-A Format (F10.0, 4I5, A8)		
Columns	Variable	Description
1-10	T1	The time for which the loads are being input (sec). .....
11-15	NCF	If concentrated ring loads are applied to the structure at time T1, set NCF = 1. If not, set NCF = 0. .....
16-20	IDELF	If distributed loads are to be applied to the shell at time T1, set IDELF = 1. If not, set IDELF = 0. .....
21-25	IDCOE	If the Fourier cosine coefficients for the distributed loadings are to be read in at time T1, set IDCOE = 1. If not, set IDCOE = 0. .....
26-30	ITCOE	If the Fourier cosine coefficients for the temperatures and temperature gradients are to be read in at time T1, set ITCOE = 1. If not, set ITCOE = 0. .....
31-38	CONSTP	If the applied loads, temperatures and temperature gradients are constant from time, T1, to the final time, TDFIME (CARD III-A), punch the word CONSTANT in columns 31-38. If these parameters are not constant, leave columns 31-38 blank. .....

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**B. Concentrated Ring Loads**

The concentrated ring loads must be specified for each harmonic. (IF NCF = 0, OMIT CARDS IX-B.)

1. Control Card - This card indicates the presence or absence of concentrated ring loads for a particular harmonic. (ONE CARD IX-B-1 FOR EACH HARMONIC.)

Card Type IX-B-1 Format (I5)		
Columns	Variable	Description
1-5	NCF1	If there are concentrated ring loads for this particular harmonic, set NCF1 = 1. If not, set NCF1 = 0. .....

2. Concentrated Ring Loads - For harmonics having ring loads associated with them, the loads are specified using these cards. (IF NCF1 = 0, OMIT CARDS IX-B-2 FOR THE HARMONIC BEING CONSIDERED.) ONE OR MORE CARDS IX-B-2 MAY BE USED, BUT NEVER UTILIZE MORE THAN 51 PER HARMONIC.

Card Type IX-B-2 Format (2I5, 4F10.0)		
Columns	Variable	Description
1-5	IN1	First node to which this loading applies. .....
6-10	IN2	Last node to which this loading applies. .....
11-20	F1	Axial ring load applied at a node (lb).* .....
21-30	F2	Circumferential ring load applied at a node (lb).* .....
31-40	F3	Radial ring load applied at a node (lb).* .....
41-50	F4	Concentrated moment applied at a node (in-lb).* .....

Examples: The use of cards IX-B should become clear after considering the following examples:

1. Consider the case where a uniform tensile ring loading of 100 psi is being applied in the axial direction to the first node of a cylinder. The solution for this problem has been presented in Figure 20 of Reference 31 The thickness of the cylinder is 0.1

-----

\* The total value of the ring load for each harmonic is input, not the load per unit length of circumference. For complicated ring loads the value of the load input for each harmonic is obtained by integrating the product of the load and the corresponding displacement function around the circumference.



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inches with the radius being given as 6 inches. Consider that harmonics 0 and 2 are being run. The total ring load for the zero harmonic will be  $(100) \times 2\pi(6) \times (0.1) = 376.9$  lb.

Five cards of type IX are required to input these loads assuming they are constant from time  $T1 = 0.0$  to  $TOTIME$  and assuming 50 elements are used to idealize the structure.

CARD	VARIABLE	VALUES
IX-A	T1 = 0.0	NCF = 1                      IDELF = IDCOE = ITCOE = 0
IX-B	NCF1 = 1	(HARMONIC 0)
IX-C	IN1=1 IN1=1	F1 = -376.9    F2 = F3 = F4 = 0
IX-C	IN1 = 2    IN1 = 51	F1 = F2 = F3 = F4 = 0
IX-B	NCF2 = 0	(HARMONIC 2)

2. The second example considers a radial ring load of  $F \cos\theta$  applied to a cylinder of radius  $r$ .

Performing the integration, one obtains the radial ring load for harmonic 1 as

$$F_3 = \int_0^{2\pi} (F \cos\theta) r \cos\theta d\theta$$

$$= \pi r F$$

The Fourier coefficients for the other harmonics are zero.

### C. Distributed Loads - (IF IDELF = 0, OMIT CARDS IX-C)

The distributed loadings may be input in one of two ways: the Fourier coefficients may be read in for each harmonic or the loadings may be specified at a desired number of circumferential angles ( $\leq 37$ ). If the second option is used, the Fourier coefficients will then be generated internally. The user should note that it is possible to input distributed loads in only one of two ways.

1. Distributed Loads - (Input at various circumferential angles). Since the choice of the displacement functions utilized in this analysis necessitate the presence of loads symmetric about the meridian  $\theta = 0$ , it is necessary to specify the distributed loadings for angles from  $0^\circ \rightarrow 180^\circ$ . The code then assumes that the distribution from  $180^\circ \rightarrow 360^\circ$  is the mirror image of the input distribution. (IF IDCOE = 1, OMIT CARDS IX-C-1)

a. Control Card - Utilize this card to indicate the number of angles for which the loads will be specified.

Card Type IX-C-1-a Format (3I5)		
Columns	Variable	Description
1-5	IELM1	First element to be distributed loading applies. .....
6-10	IELM2	Last element to which this distributed loading applies. .....
11-15	NDP	Number of circumferential angles at which the distributed loads are to be specified ( $1 \leq NDP \leq 37$ ). If the loadings are constant in the circumferential direction set NDP = 1. .....

b. Distributed Loads at Specified Angles\* This card specifies the angle at which the loads are being input and provides the values of the loads at that angle. (INCLUDE NDP CARDS OF

\* The first loading must always be given for  $\theta = 0^\circ$ . The next loading is given at the angle where the load changes in value. If the load is constant with respect to  $\theta$ , only one card will be necessary to input the load. Do not input values for the loads at  $\theta = 180^\circ$  since the load at that angle will be equal in all cases to the load input at the previous value of THETAB.

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TYPE IX-C-1-b FOR EACH CARD IX-C-1-a.)

Card Type IX-C-1-b Format (4F10.0)		
Columns	Variable	Description
1-10	THETB	Circumferential angle (degrees) for which this data is given. .....
11-20	P	Distributed load in the meridional direction (psi). .....
21-30	R	Distributed load in the normal direction (psi). .....
31-40	S	Distributed load in the circumferential direction (psi). .....

Example: Consider the normal pressure distribution on an element depicted in Figure 4. To input the pressure on this element requires specification of the pressures for four values of  $\theta$ .

THETB	R (I)
0.0	-Q1
30.0	-Q2
90.0	-Q3
2.0	0.0

2. Distributed Loads - (Fourier Coefficients) The Fourier coefficients for the distributed loads may be specified using these cards. The coefficients must be specified (even though they may be zero) for each harmonic being employed in the analysis. The coefficients are specified for each harmonic of the first group of elements, then for each harmonic of the second group, etc. until the values have been input for all the elements. (IF IDCOE = 0, OMIT CARDS IX-C-2)

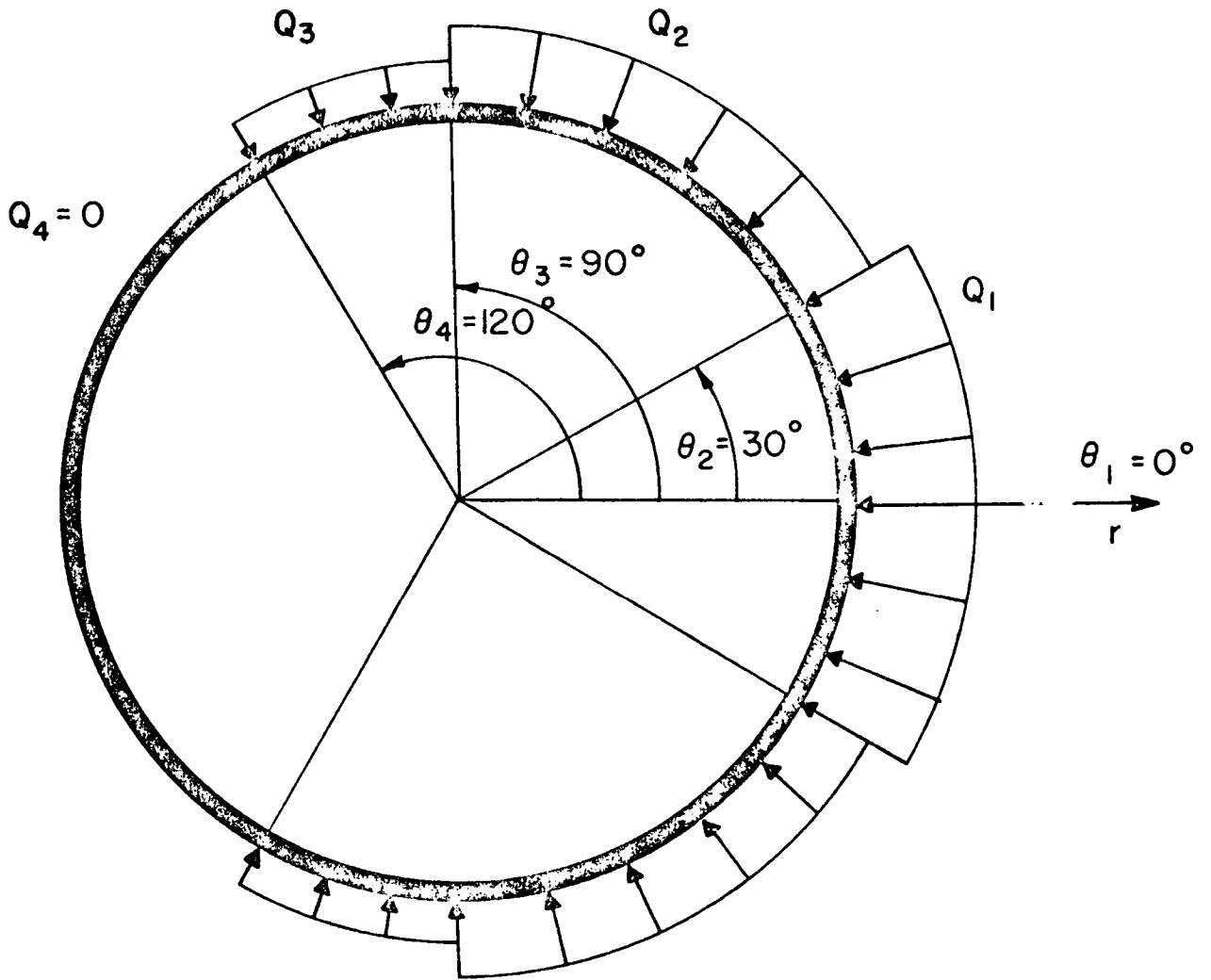


FIG 4 REPRESENTATIVE VARIATION OF DISTRIBUTED LOADS APPLIED TO A TYPICAL ELEMENT

a. Control Card

Card Type IX-C-2-a Format (2I5)		
Columns	Variable	Description
1-5	IELM1	First element to which these loads apply. .....
6-10	IELM2	Last element to which these loads apply. .....

b. Fourier Coefficients - (NH CARDS OF TYPE IX-C-2-b FOR EACH CARD IX-C-2-a.)

Card Type IX-C-2-b Format (3F10.0)		
Columns	Variable	Description
1-10	P	Fourier coefficient of the distributed load in the meridional direction for a particular harmonic (psi). .....
11-20	R	Fourier coefficient of the distributed load in the normal direction for a particular harmonic (psi). .....
21-30	S	Fourier coefficient of the distributed load in the circumferential direction for a particular harmonic (psi). .....

D. Temperature Distribution and Gradients

Essentially the same logic is employed for inputting the temperatures and gradients that was used for the specification of the distributed loads. The explanation of this procedure should therefore not need be repeated.

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The temperatures are specified for the midsurface of the shell. The temperature gradients (through the thickness) are considered positive if the temperature for the outer surface is greater than the temperature on the inner surface. (IF ITELF = 0, OMIT CARDS IX-D.)

1. Temperature Distribution and Gradients - (Input at various circumferential angles)

Again, the requirement of symmetry about the meridian  $\theta = 0$ , makes it necessary to specify the temperature distribution and thermal gradients only from  $0^\circ \rightarrow 180^\circ$ . The temperature distribution and gradients are input on the same cards for the various angles. (IF ITCOE = 1, OMIT CARDS IX-D-1.)

- a. Control Card - Utilize this card to indicate the number of angles for which the temperature and gradients will be specified.

Columns	Variable	Description
1-5	IELM1	First element to which this data applies. .....
6-10	IELM2	Last element to which this data applies. .....
11-15	NDP	Number of circumferential angles at which the temperature distribution and gradient are to BE SPECIFIED ( $1 \leq NDP \leq 37$ ). If the temperature is constant in the circumferential direction, set NDP = 1. .....

b. Temperature and Temperature Gradient at Specified Angles -

This card specifies the angle at which the temperature and temperature gradient (through the thickness) is being input and provides the value of the temperature at that angle. (INCLUDE NDP CARDS OF TYPE IX-D-1b

FOR EACH CARD IX-D-1-a.)

Card Type IX-D-1-b Format (3F10.0)		
Columns	Variable	Description
1-10	THETB	Circumferential angle for which this temperature and gradient are given. .....
11-20	P	Distributed temperature at $\theta = \text{THETB}$ ( $^{\circ}\text{F}$ ). .....
21-30	R	Temperature gradient (through the thickness) at $\theta = \text{THETB}$ ( $^{\circ}\text{F}/\text{in}$ ). .....

2. Temperature Distribution and Gradient - (Fourier Coefficients)

If the user so desires, the Fourier coefficients for the temperature distribution and gradient may be specified for each of the harmonics being used. Again, the coefficients are specified for all harmonics for the first group of elements, then for the second group, etc., until all the element coefficients have been input. (IF ITCOE = 0, OMIT CARDS IX-D-2)

a. Control Card

Card Type IX-D-2-a Format (2I5)		
Columns	Variable	Description
1-5	IELM1	First element to which these properties apply. .....
6-10	IELM2	Last element to which these properties apply. .....

b. Fourier Coefficients - (NH CARDS OF TYPE IX-D-2-b FOR EACH

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CARD IX-D-2-a.)

Card Type IX-D-2-b Format (2F10.0)		
Columns	Variable	Description
1-10	TH1	Fourier coefficient of the temperature distribution (°F) for a particular harmonic. .....
11-20	DTH1	Fourier coefficient of the temperature gradient (°F/in) for a particular harmonic. .....

X. FINAL DATA CARD FOR A CASE

Place this card after the last card IX of each data set. This signifies the end of the input data for a case. (ONE CARD X PER DATA SET.)

Card Type X	
Columns	Punch
1-11	END OF CASE .....

XI. FINAL DATA CARD FOR A RUN

This card must be placed after the card X of the last case to be run.



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It denotes the end of the input data for a run. (ONE CARD XI PER RUN)

Card Type XI	
Columns	Punch
1-10	END OF RUN .....

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## SECTION IV

## EXAMPLE PROBLEMS

The example problems which follow were chosen to demonstrate the versatility of the code and to further acquaint the users with the procedures for inputting the data to the code. The data presented herein is typical for the problems solved by the code and demonstrates many of the input procedures.

Since the most complex portion of the input data is the specification of the loading conditions, a variety of loadings are demonstrated. Response curves are presented so the user may check his output with the previously obtained curves. The first two example problems utilize the shells described in example problems 1 and 2 of the SAMMSOR user's guide (Ref. 1) while the third example problem demonstrates the two procedures for specifying distributed pressure loadings.

## Example Problem 1

The first example problem was chosen to demonstrate the procedure for inputting a concentrated ring load and to demonstrate the program's capability to solve highly nonlinear problems. For the forty pound load applied in this problem, the static solution shows that the nonlinear displacement is more than four times as large as the linear solution.

The shell to which the load is applied is the shallow spherical cap ( $\lambda=6$ ) utilized in the first example problem in the SAMMSOR user's guide. The edges of the shell are assumed to be clamped. Since the loading is symmetric, the displacements and stresses will be calculated only along the line  $\theta = 0$ . Only the response for the zeroth harmonic will be determined. A set of input data for this case is presented in Figure 5 with the displacement response of the apex of the shell being presented in Figure 6. This response curve should allow the user to check his version of the code.

## Example Problem 2

The shell described in the second example problem in the SAMMSOR user's guide is now subjected to a 50 psi internal pressure. The load-in is applied at time  $T1 = 0.0$  and remains constant for the duration of the calculation.

Two sets of input data are provided for this example problem. The first set (Figure 7) allows the program to calculate the response for the first 300 time steps. The second set of input data (Figure 8) will

NCASE= 1

PRINTOUT OF INPUT DATA

Fig. 5 INPUT DATA - EXAMPLE PROBLEM 1

CARD TYPE	10	20	30	40	50	60	70	80
II - A	6	4						
- B	*****							
- B	EXAMPLE PROBLEM NO. 1				DYNASOR II USER'S MANUAL			
- B	THE SHELL DESCRIBED IN EXAMPLE PROBLEM 1 OF THE SAMMSOR USER'S GUIDE							
- B	IS SUBJECTED TO A 40 LB. APEX LOADING WITH THE SOLUTION BEING DETERMINED							
- B	FOR 400 TIME STEPS							
- B	*****							
III - A	0.0001	.00000025	0	0	1	0		
- B	1	4	1	8	1	100	1	1
IV - A	1							
- B		0.0						
V	1	0						
VI - A	4							
- B	31	1						
- B	31	2						
- B	31	3						
- B	31	4						
VII - A	0	0						
IX - A	0.0	1	0	0	0	0	0	0
- B	- 1	1						
- 2	1	1	40.0	0.0	0.0	0.0	0.0	0.0
- 2	2	31	0.0	0.0	0.0	0.0	0.0	0.0
X	END OF CASE							

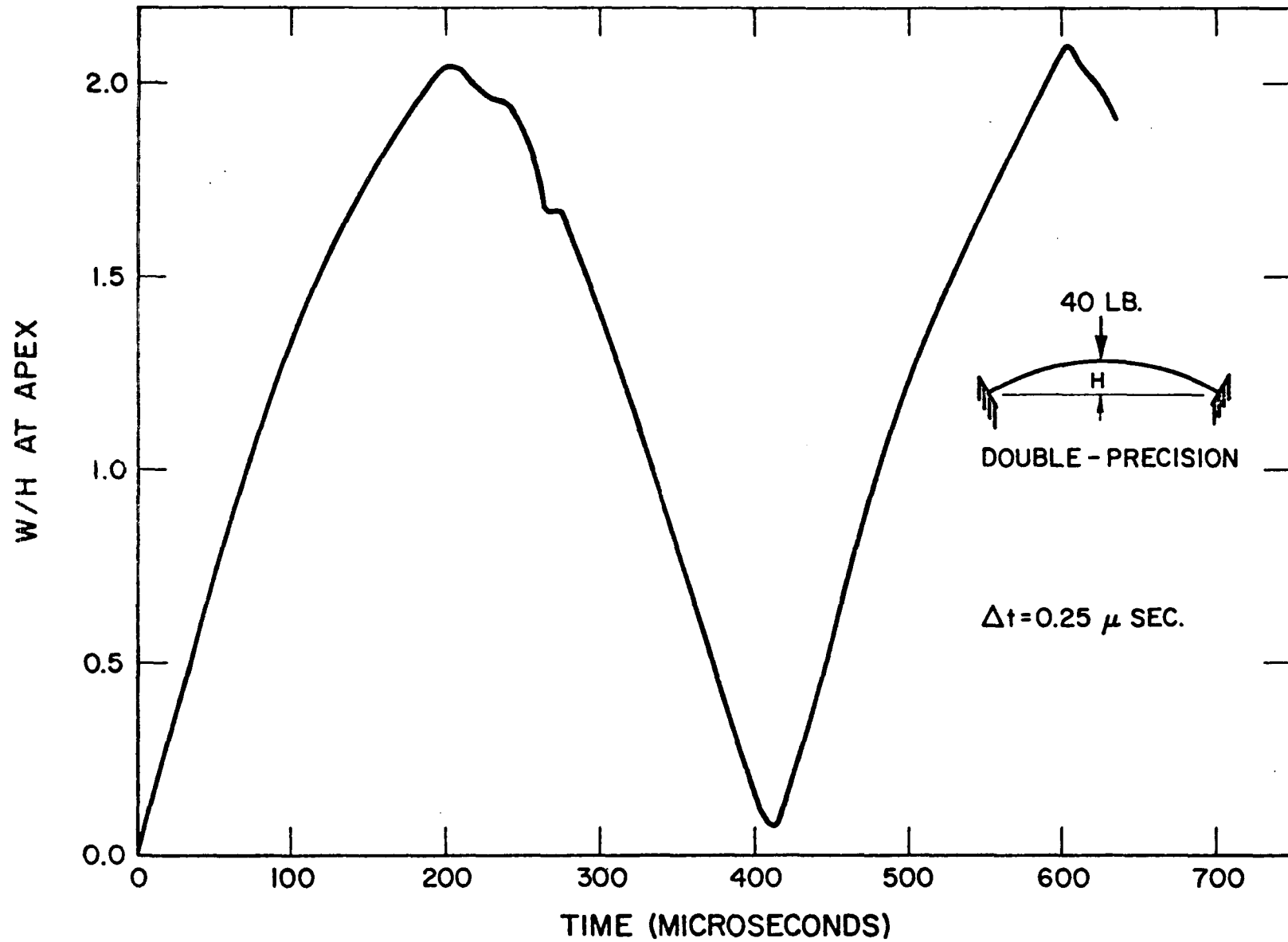


FIG. 6 APEX DISPLACEMENT RESPONSE UNDER CONCENTRATED AXIAL LOAD

NCASE= 2

PRINTOUT OF INPUT DATA

Fig. 7 INPUT DATA - EXAMPLE PROBLEM 2

CARD	10	20	30	40	50	60	70	80
TYPE	12345678901	2345678901	2345678901	2345678901	2345678901	2345678901	2345678901	234567890
II - A	6	8						
- B	*****							
- B	EXAMPLE PROBLEM NO. 2				DYNASOR II USER'S MANUAL			
- B	CAP-TORUS-CYLINDER CONFIGURATION							
- B	THE SHELL DEPICTED IN THE SECOND EXAMPLE PROBLEM OF THE SAMSOR USER'S							
- B	MANUAL IS SUBJECTED TO A 50 PSI INTERNAL PRESSURE							
- B	*****							
III - A	0.0009	0.00003	0	0	1	0		
- B	1	10	1	20	1	100	1	0 1
IV - A	1							
- B		0.0						
V	1	0						
VI - A	4							
- B	51	1						
- B	51	2						
- B	51	3						
- B	51	4						
VII - A	0	0						
IX - A		0.0	0	1	0	OCONSTANT		
- C	-1-a	1	50	1				
	-b	0.0	0.0	50.0	0.0			
X	END OF CASE							

NCASE= 3

PRINTOUT OF INPUT DATA

CARD	10	20	30	40	50	60	70	80
TYPE	12345678901	2345678901	2345678901	2345678901	2345678901	2345678901	2345678901	2345678901
II - A	6	8						
- B	*****							
- B	EXAMPLE PROBLEM NO. 2				DYNASOR II USER'S MANUAL			
- B	THE INPUT DATA NECESSARY TO RESTART THE CODE AT TIME INCREMENT 300							
- B	IS PROVIDED TO GUIDE THE USER IN HIS RESTART OPERATIONS. THE PROBLEM							
- B	IS TO BE RUN FOR AN ADDITIONAL 300 TIME INCREMENTS.							
- B	*****							
III - A	0.0018	0.000003	1 300	1 0				
- B	1 10	1 20	1 100	1 1	0	0		
IV - A	1							
- B	0.0							
IX - A	0.0009	0 1	0 0	CONSTANT				
- C	- 1 -a 1 50	1						
	-b 0.0	0.0	50.0	0.0				
X	END OF CASE							

Fig. 8 INPUT DATA - EXAMPLE PROBLEM 2 - RESTART MODE

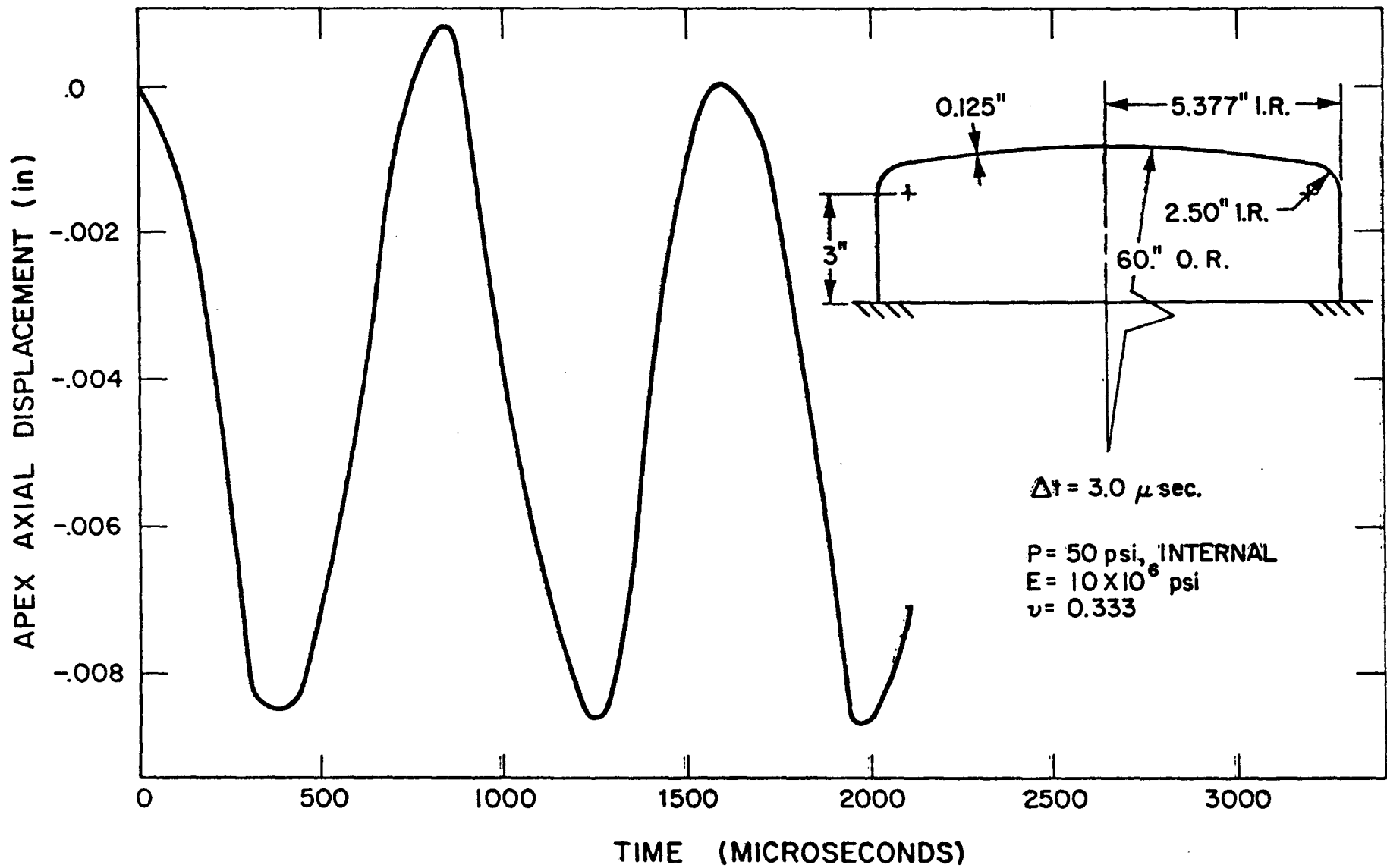


FIG. 9 DISPLACEMENT RESPONSE UNDER INTERNAL PRESSURE

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restart the code at the end of the 300th time step and will then allow the program to calculate the response for an additional 300 increments.

Since this problem is only moderately nonlinear, it is interesting to note that a much larger time step can be used for this problem than was employed in the previous example problem. The displacement response obtained for this problem is presented in Figure 9.

### Example Problem 3

This example problem was selected to demonstrate the procedures for inputting the distributed loadings on a shell. A cylindrical shell (figure 10) is subjected to a half cosine loading which is symmetric about the meridian  $= 0$ . This load is applied along the entire length of the shell. The pressure loading may be specified in one of two ways:

- 1) The Fourier coefficients may be input for each harmonic.
- 2) The pressure may be specified at various circumferential angles with the Fourier coefficients then being internally generated.

The first set of input data (Figure 11) utilizes the first of the above procedures and inputs the Fourier coefficients. The input data presented in Figure 12 describes the loading by specifying the value of the pressure at the various angles. The same procedure is employed to describe the temperature and temperature gradient distributions.

Considering the symmetry of the loading and the boundary conditions applied to this shell, it can easily be recognized that the displacements and stresses will be symmetric about the center of this cylindrical tube. Therefore, only one-half of the shell needs to be analyzed. The plane of symmetry is assured by applying an axial and a rotational restraint at node one (1).



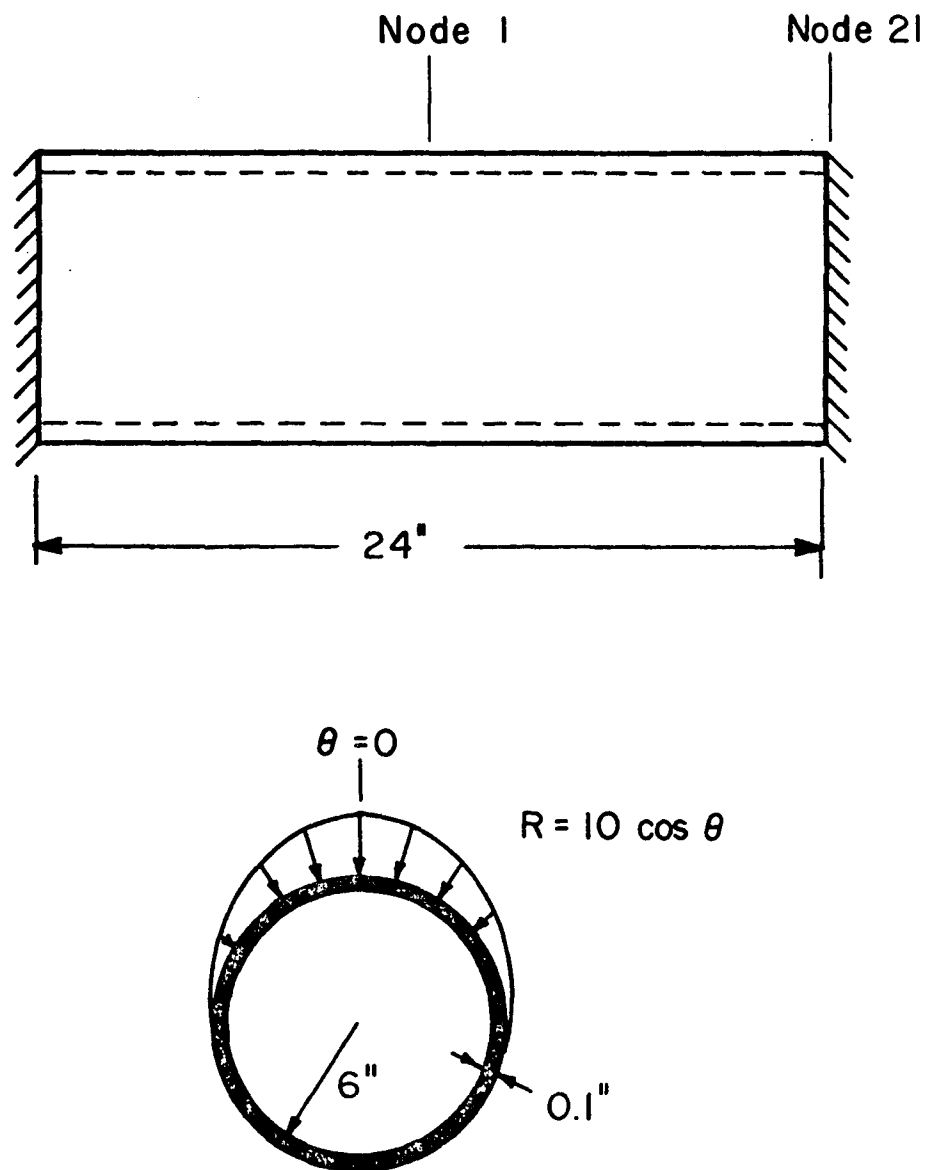


FIG 10 CYLINDRICAL SHELL SUBJECTED TO HALF COSINE PRESSURE LOADING

NCASE= 5

PRINTOUT OF INPUT DATA

Fig. 11 INPUT DATA - (SET #1) - EXAMPLE PROBLEM 3

CARD TYPE	10	20	30	40	50	60	70	80
	12345678901	2345678901	2345678901	2345678901	2345678901	2345678901	2345678901	2345678901
II - A	6	4						
- B	*****							
- B	EXAMPLE PROBLEM NO. 3				DYNASOR II USER'S MANUAL			
- B	CYLINDRICAL SHELL IDEALIZED USING 30 ELEMENTS IS SUBJECTED TO A HALF COSINE							
- B	LOADING TO DEMONSTRATE THE OPTIONS FOR INPUTTING DISTRIBUTED LOADS.							
- B	** IN THIS CASE THE PRESSURE IS SPECIFIED BY INPUTTING THE FOURIER COEFFICIENTS							
- B	*****							
III - A	0.0005	0.00001	0	0	0	0		
- B	1	5	1	10	1	50	1	1 0 1
IV - A	2							
- B		0.0		30.0				
V - A	5	0	1	2	3	4		
VI - A	6							
- B	1	1						
- B	1	4						
- B	21	1						
- B	21	2						
- B	21	3						
- B	21	4						
VII - A	0	0						
IX - A		0.0	0	1	1			OCONSTANT
- C - 2-	a1	20						
- b	0.0	-3.1831			0.0			
- b	0.0	-5.0000			0.0			
- b	0.0	-2.1221			0.0			
- b	0.0	0.0000			0.0			
- b	0.0	0.4244			0.0			
X	END OF CASE							

NCASE= 4

PRINTOUT OF INPUT DATA

CARD	10	20	30	40	50	60	70	80
TYPE	12345678901	2345678901	2345678901	2345678901	2345678901	2345678901	2345678901	2345678901
II - A		6	4					
- B		*****						
- B		EXAMPLE PROBLEM NO. 3 DYNASOR II USER'S MANUAL						
- B		CYLINDRICAL SHELL IDEALIZED USING 30 ELEMENTS IS SUBJECTED TO A HALF COSINE						
- B		LOADING TO DEMONSTRATE THE OPTIONS FOR INPUTTING DISTRIBUTED LOADS.						
- B		** IN THIS CASE THE PRESSURE IS SPECIFIED AT VARIOUS CIRCUMFERENTIAL ANGLES **						
- B		*****						
III - A		0.0005	0.00001	0	0	0	0	
- B		1	5	1	10	1	50	1 1 0 0
IV - A		2						
- B			0.0	30.0				
V		5	0	1	2	3	4	
VI - A		6						
- B		1	1					
- B		1	4					
- B		21	1					
- B		21	2					
- B		21	3					
- B		21	4					
VII - A		0	0					
IX - A		0.0	0	1	0	OCONSTANT		
- C - 1 - a		1	20	37				
- b			0.0		0.0	- 9.9976		0.0
- b			2.5		0.0	- 9.9786		0.0
- b			5.0		0.0	- 9.9406		0.0
- b			7.5		0.0	- 9.8836		0.0
- b			10.0		0.0	- 9.8079		0.0
- b			12.5		0.0	- 9.7134		0.0
- b			15.0		0.0	- 9.6005		0.0
- b			17.5		0.0	- 9.4693		0.0
- b			20.0		0.0	- 9.3201		0.0
- b			22.5		0.0	- 9.1531		0.0
- b			25.0		0.0	- 8.9687		0.0
- b			27.5		0.0	- 8.7673		0.0
- b			30.0		0.0	- 8.5491		0.0
- b			32.5		0.0	- 8.3147		0.0
- b			35.0		0.0	- 8.0644		0.0
- b			37.5		0.0	- 7.7988		0.0
- b			40.0		0.0	- 7.5184		0.0
- b			42.5		0.0	- 7.2236		0.0
- b			45.0		0.0	- 6.9151		0.0
- b			47.5		0.0	- 6.5935		0.0
- b			50.0		0.0	- 6.2592		0.0
- b			52.5		0.0	- 5.9131		0.0
- b			55.0		0.0	- 5.5557		0.0
- b			57.5		0.0	- 5.1877		0.0
- b			60.0		0.0	- 4.8059		0.0
- b			62.5		0.0	- 4.4229		0.0
- b			65.0		0.0	- 4.0275		0.0
- b			67.5		0.0	- 3.6244		0.0
- b			70.0		0.0	- 3.2144		0.0
- b			72.5		0.0	- 2.7983		0.0
- b			75.0		0.0	- 2.3769		0.0
- b			77.5		0.0	- 1.9509		0.0
- b			80.0		0.0	- 1.5212		0.0
- b			82.5		0.0	- 1.0887		0.0
- b			85.0		0.0	- 0.6540		0.0
- b			87.5		0.0	- 0.2181		0.0
- b			90.0		0.0	0.0000		0.0
X		END OF CASE						

Fig. 12 INPUT DATA - (SET #2) - EXAMPLE PROBLEM 3

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## Appendix 5 - Use of the Restart Option

In order for efficient use to be made of the DYNASOR II code, the user should become familiar with the option provided for restarting the program. Through effective use of this option the dynamic response studies can be completed using a minimum amount of computer time.

Use of the restart option may prove invaluable in a number of situations. Abnormal termination of the program may occur if a numerical instability is noted in the response. If this occurs, the restart option can be used with a different value of the time increment. Another important use of the restart option arises when the user is satisfied with the results previously obtained but desires to extend the response data to a further point in time. In such a case the program is restarted at the last time step for which the restart information was placed on tape. A most effective use of this option can be made when conducting dynamic stability analyses where it is desirable to evaluate the response to see if buckling has occurred. If it has not, the decision can then be made to extend the run to further points in time.

Utilizing large time steps can result in a damping effect upon the solution so it is advisable to run the problem for a couple of oscillations, check to see if the solution is significantly damped, and then run the problem for the desired number of oscillations. If an evaluation of the initial results indicates that a smaller or larger time step should be used, the restart facility might be used to keep from having to repeat the initial calculations.

The displacements, velocities, and forces should be written on tape for almost all of the cases to insure that the restart information will be available if an evaluation of the calculated response indicates that the program should be restarted. The time required to write the restart information on tape is negligible when compared with the amount of time required to obtain the total response.

If it is desirable to decrease the time increment when restarting the program, the user should exercise care in selection the increment (INRST) at which the program will be restarted. The decision to decrease the size of the time step will usually be based upon the observation that the solution has become unstable or that significant damping is present in the response. To restart the program the user must be sure that the increment (INCRST) has been selected small enough to insure that the inaccuracies created by the larger time step can be neglected.

On the other hand, if the results from a previous run indicate that it is possible to increase the size of the time step for the remaining calculations, then care must also be taken in the selection of INCRST. For the numerical extrapolation procedure to produce accurate sets of displacements, it is recommended that the solution be restarted on a

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relatively straight portion of the displacement response curve. Considering the curve presented in Figure 6, it would be recommended that the program be restarted at 500 microseconds rather than at 600 microseconds because of the extrapolation procedure being utilized (i.e. the curve is smoother at 500 microseconds).

When using the restart option, it is possible to specify different values for a number of the control constants and input parameters. The data on cards I-IV may be changed, but the same Fourier harmonics and boundary conditions must be used. It is also required that the coefficients of thermal expansion remain the same when restarting the program. These requirements allow the user to omit card types V, VI, and VII when preparing data for restart operations. The considerations effecting the input of the loads for restart operations are presented in Appendix 6.

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## Appendix 6 - Load and Temperature Input Discussion

Since the DYNASOR II program accepts time varying loading and temperature conditions, the logic required to input these conditions is of necessity more complex than the logic required to input the other parameters. A discussion of the procedures for inputting these loading conditions is contained in this section. In this appendix the term loads refers to all distributed and concentrated forces while the term temperatures refers to both the temperature and temperature gradient distributions.

If there are no loads or no temperatures, it should be noted that a proper selection of the input constants allows omission of the input cards pertaining to the missing terms. In other words, the user selects the proper values for input keys and the proper read statements are automatically skipped.

To illustrate the procedure for inputting time varying loads and temperatures the information presented in Figure A6-1 is utilized. The load-time and temperature-time curves are approximated as a series of linear segments by specifying values of both the loads and temperatures at discrete points in time and then assuming linear variations between the times. In order to specify the loads and temperatures in Figure A6-1, it is necessary to specify both the loads and the corresponding temperatures at times  $T_1$ ,  $T_2$ , and  $T_3$ . Both the applied loads and temperatures are constant from time  $T_3$  to the selected TOIME so the value of CONSTF should be set equal to CONSTANT at time  $T_3$ . Obviously, if the loads or temperatures vary rapidly with time, it may be necessary to specify these conditions at a large number of times in order for the linear variation to be an accurate representation of the load-time and temperature-time curves.

The logic for the load and temperature input is now discussed for each of the two program start conditions, namely:

IRSTRT = 0 Calculation begins at time increment = 0

IRSTRT = 1 Calculation begins at time increment = INCRST

1. The loads and the temperatures must both be input at each time  $T_i$  at which the loads or temperatures vary. In other words, the loads cannot be input at one time and the temperatures at another.
2. The difference between successive times at which the loads and temperatures are input ( $T_{i+1} - T_i$ ) must always be

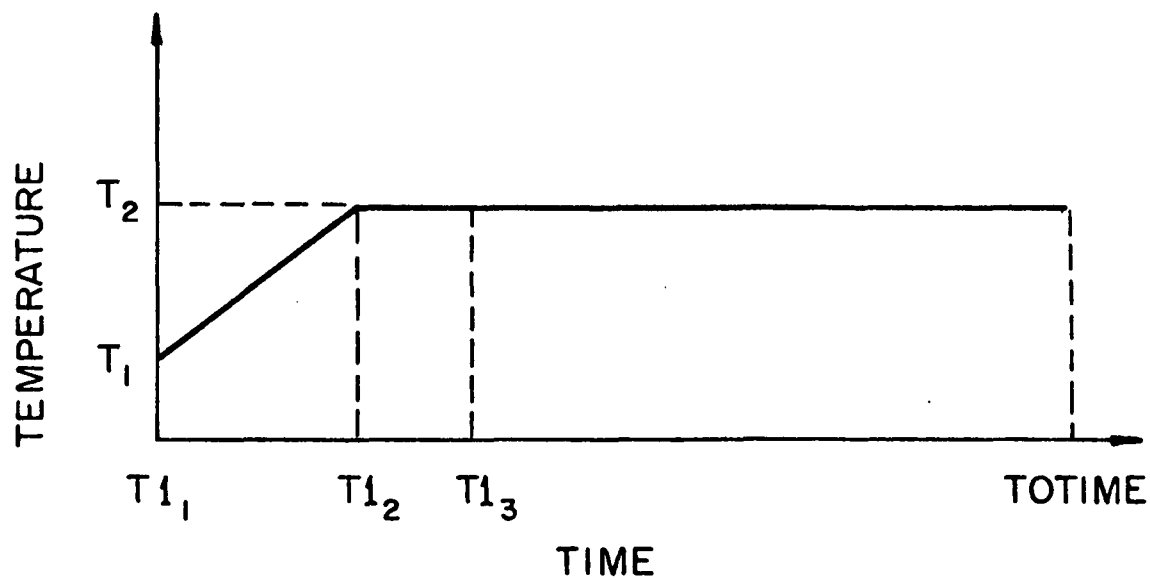
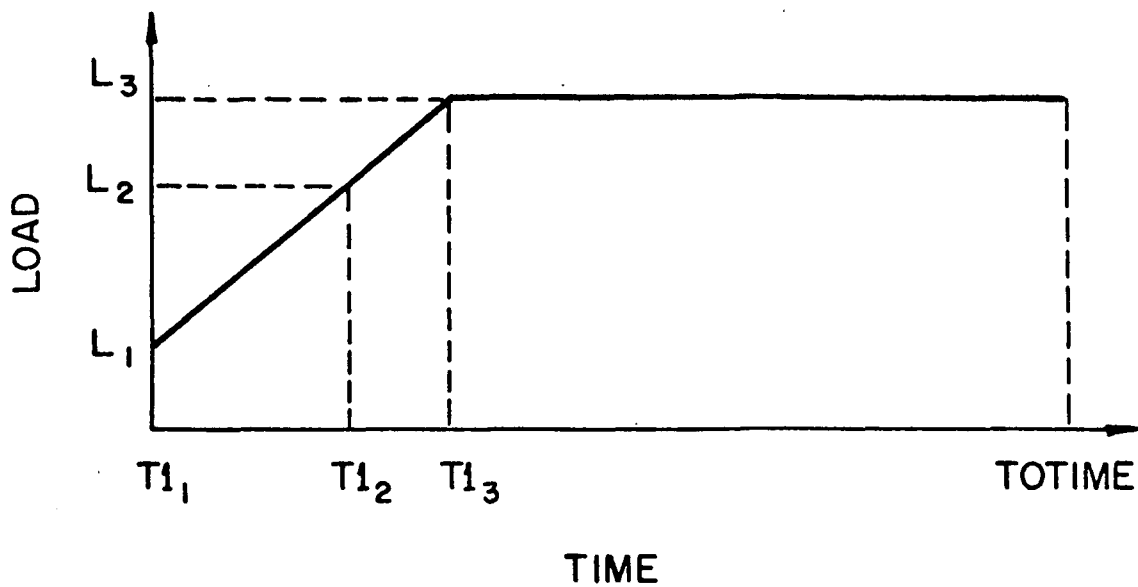


FIG A6-1 MECHANICAL AND THERMAL LOAD HISTORY FOR AN ELEMENT



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greater than the time increment (DELTE) specified for solving the equations of motion.

INRART = 0

The cases which may arise when considering the loads and temperatures and the input logic required to describe these situations are as follows when the program is making an initial run on a problem:

CASE	INPUT LOGIC
<p>1. Loads and temperatures and constant (in time) on each element. Note, however, that variations from element to element are allowed.</p>	<p>Input only one set of loads and temperatures. These must be specified at time <math>T1 = 0.0</math> and the value of CONSTP should be read as CONSTANT.</p>
<p>2. Loads or temperatures (or both) vary with time.</p>	<p>Input, in order, the loads and temperatures at times <math>T1_1</math> (must be equal to 0.0), <math>T1_2</math>, <math>T1_3</math>, ... until the value of <math>T1</math> reaches or exceed the value of TOTIME (total time for the case.) The columns for CONSTP should be left blank.</p>

IRSTRT = 1

The program may be restarted utilizing a new value for TOTIME which may be less than, equal to, or greater than the value which was utilized in the previous run which created and stored the restart information for use in this run. The previous value of TOTIME will be referred to as TOTIMEP. The input logic varies according to the relative values of TOTIME and TOTIMEP so each possible combination will be discussed separately.

Procedures which may not be utilized in the restart mode are:

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1. If the program was originally run as case A with IRSTRT = 0, it is not possible to input loads and temperatures at any time until the value of TOTIMEP has been exceeded.
2. Consider that the program is being restarted at a time which is within the interval  $T1_j$  --  $T1_{j+1}$ . The loads and temperatures were input in the previous run for times  $T1_j$  and  $T1_{j+1}$ . The first value of  $T1$  for which the loads and temperatures may be specified in the restart mode must be greater than the time  $T1_{j+1}$  which was utilized in the previous run.

Consideration will first be given to the cases where the new value of the maximum time is less than or equal to the one previously used.

TOTIME ≤ TOTIMEP

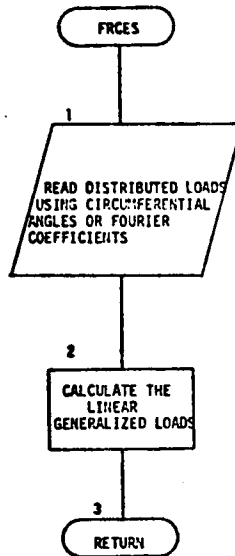
CASE	INPUT LOGIC
1. Both the loads and the temperatures are constant (in time) and are equal to the values specified for IRSTRT = 0, Case 1.	No loads or temperatures are input
2. Both the loads and temperatures are constant (in time) but are different from the values specified for IRSTRT = 0, Case 1.	This problem is not allowed by the program. If the user desires to run this case, it is suggested that the problem be rerun beginning at time = 0.0.
3. Loads or temperatures vary with time. (This cannot be a restart of Case 1, IRSTRT = 0.)	Input loads and temperatures at times $T1_1, T1_2, \dots$ until the value $T1_n$ reaches or exceeds the value of TOTIME. The value of $T1$ must be greater than the value of $T1_{j+1}$ of the previous run

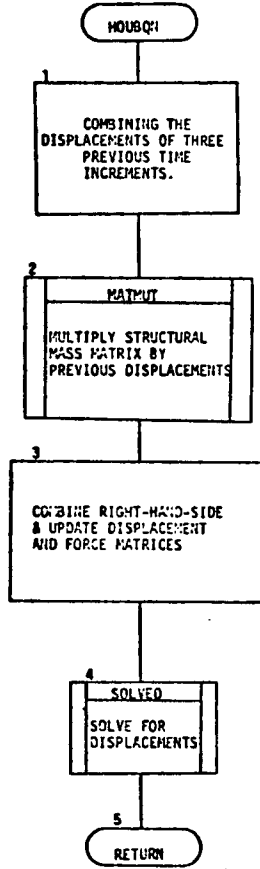
October 1972

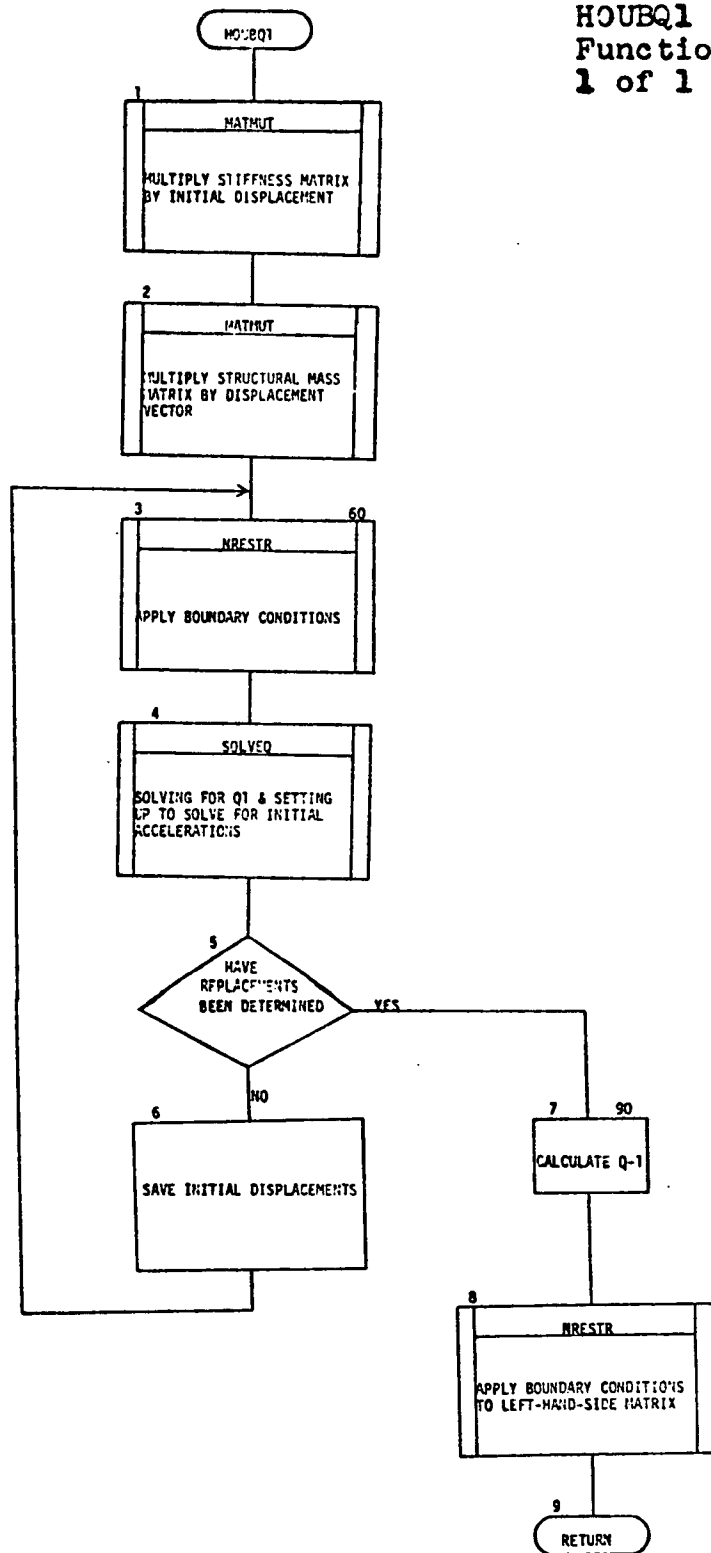
The possible cases which may arise if the value of TOTIME is greater than TOTIMEP are now presented. It should be noted that cases differ only slightly from those previously discussed.

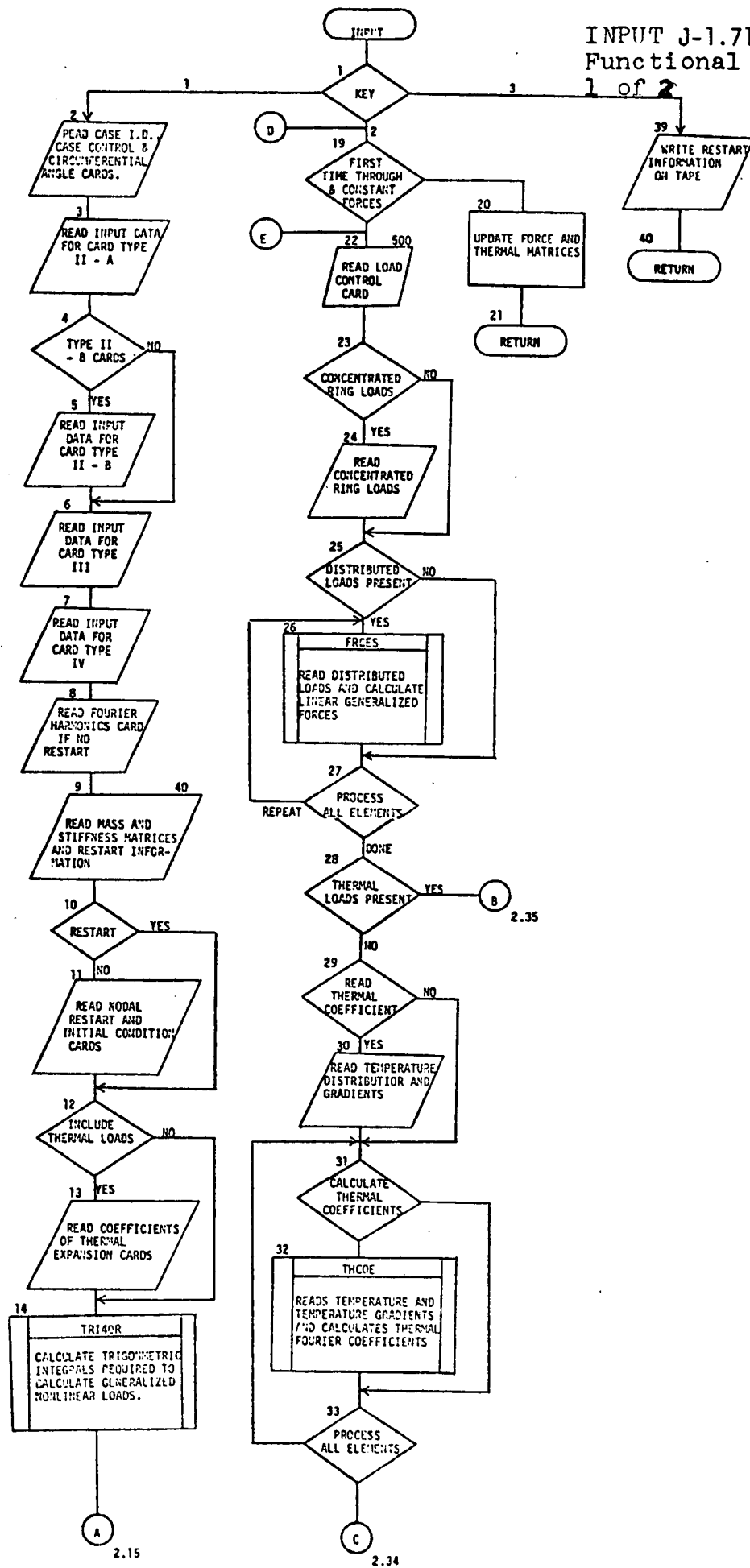
## TOTIME &gt; TOTIMEP

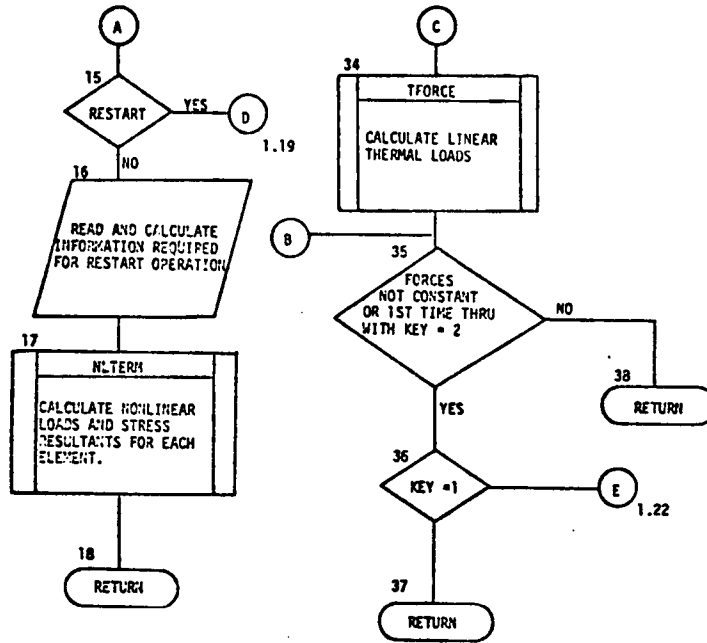
CASE	INPUT LOGIC
1. Both the loads and temperatures are constant (in time) and are equal to the values specified for IRSTRT = 0, Case 1.	The loads and temperatures must be input for T1 = TOTIMEP and the value of CONSTP is set as CONSTANT. The specified loads and temperatures must be identical with those read for the previous run (IRSTRT = 0).
2. Both the loads and temperatures are constant (in time) but are different from the values specified for IRSTRT = 0, Case 1.	The new loads will not be applied until TOTIMEP is reached. The logic for Case 1, above, is then applied.
3. Loads at temperatures (or both) vary with time.	The loads and temperatures must be input at times T1 <sub>1</sub> , T1 <sub>2</sub> , ... until the value T1 <sub>n</sub> reaches or exceeds the value of TOTIME.



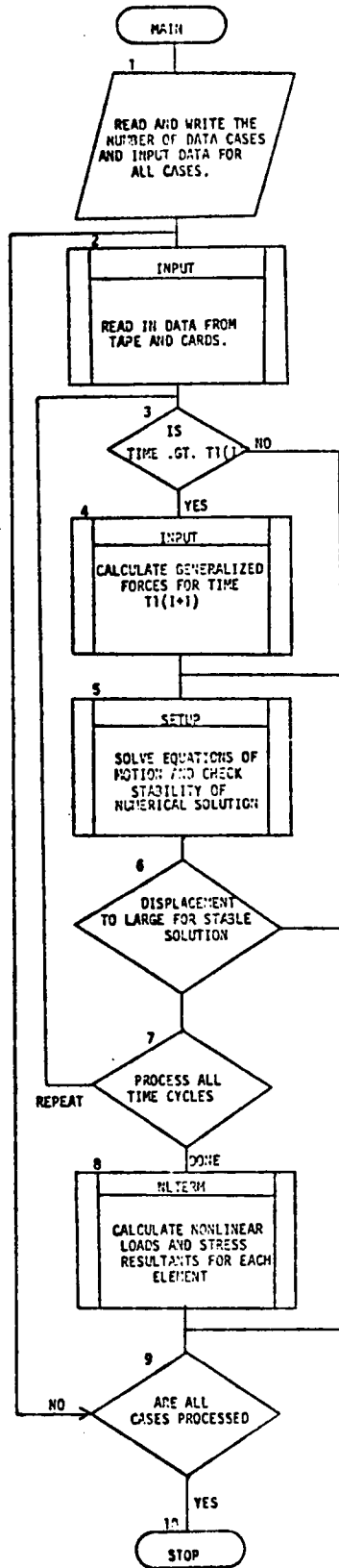


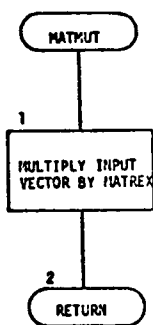


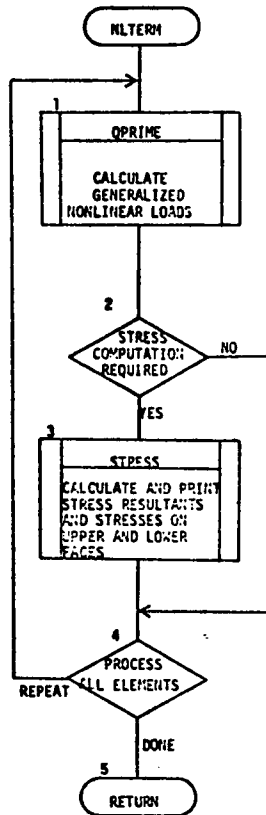


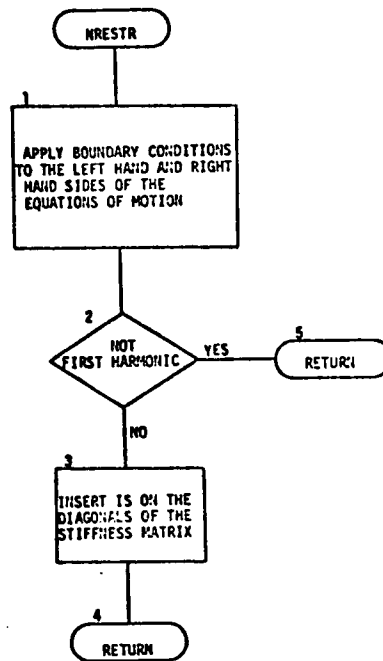


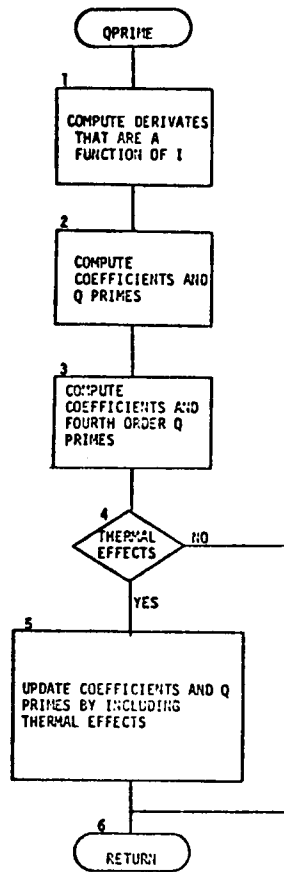


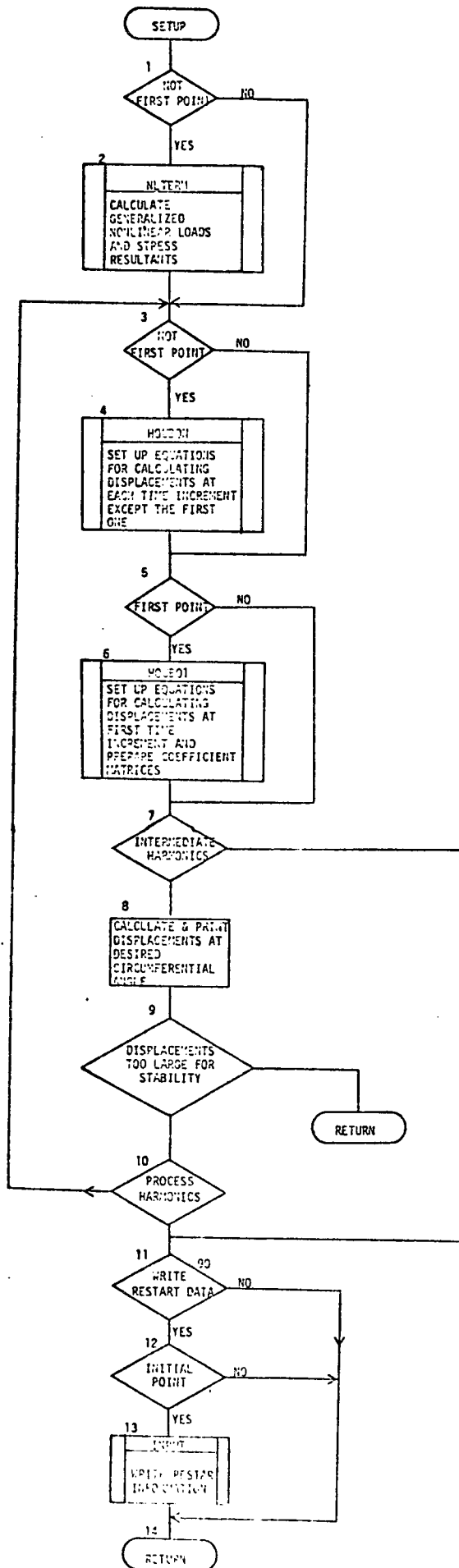


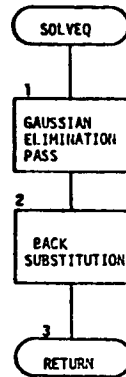


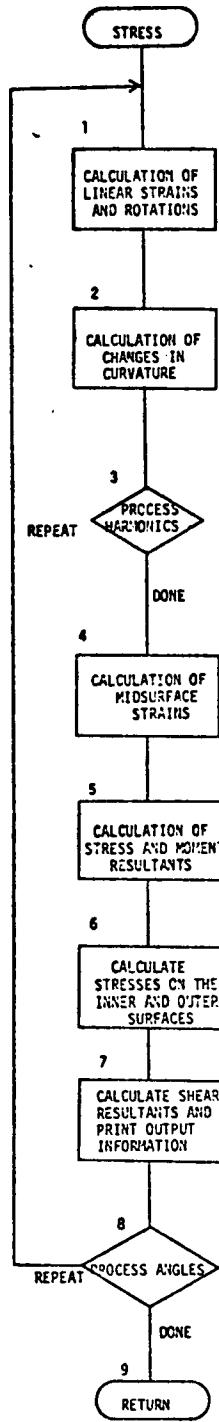




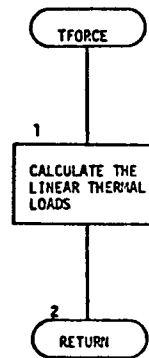


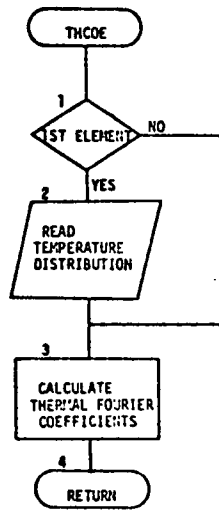


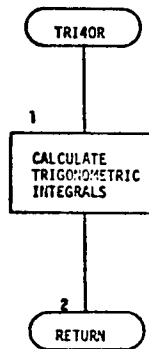












DYNASOR II - A FINITE ELEMENT PROGRAM FOR THE  
DYNAMIC NONLINEAR ANALYSIS OF SHELLS OF REVOLUTION

Joe R. Tillerson and Walter E. Haisler

OPERATION MANUAL

October 15, 1970

TEES-RPT-70-19

Texas A&M University  
College Station, Texas

J-2-1

## ABSTRACT

The DYNASOR II program is used for the Dynamic Nonlinear Aalysis of Shells Of Revolution. The equations of motion of the shell are solved using Houbolt's numerical procedure. The displacements and stress resultants can be determined for both symmetrical and asymmetrical loading conditions. Asymmetrical dynamic buckling can be investigated. Solutions can be obtained for highly nonlinear problems utilizing as many as five of the harmonics generated by SAMMSOR program. A restart capability allows the user to restart the program at a specified time.

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	H. Checkpoint, Restart, Error Procedures, Backup, Recovery Procedures . . . . .	J-2.24

## SOR - Shell Of Revolution

## Computer Programs

A family of compatible computer codes for the analysis of the shell of revolution (SOR) structures has been developed by researchers at Texas A&M University. These analyses employ the matrix displacement method of structural analysis utilizing a curved shell element. Geometrically nonlinear static and dynamic analyses can be conducted using these codes. The important natural frequencies and mode shapes can also be determined by employing another of the codes. Efficient programming provides codes capable of performing these desired analyses in relatively small amounts of computer time.

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SAMMSOR II - Stiffness And Mass Matrices for Shells Of Revolution are generated utilizing the first member of this family. This program accepts a description of the structure in terms of the coordinates and slopes of the nodes and the properties of the elements joining the nodes. For shells with simple geometries (such as cylinders, shallow caps, hemispheres, etc.) the shell geometry can be internally generated. Utilizing the element properties, the structural stiffness and mass matrices are generated for as many as twenty harmonics and stored on magnetic tape. Each of the other SOR programs utilizes the output tape generated by SAMMSOR as input data for the respective analyses. One advantage of creating the stiffness and mass matrices in a separate program is that a variety of analyses can be performed on the same shell configuration without having to create the matrices more than once. Obviously, a variety of boundary and loading conditions can be employed without having to create new mass and stiffness matrices for each case.

SNASOR II - The Static Nonlinear analysis of Shells Of Revolution subjected to arbitrary mechanical and thermal loading is performed using the second computer code. Utilizing the stiffness matrices generated by SAMMSOR and the loading conditions and boundary conditions input to SNASOR II, the equilibrium equations for the structure are generated. The nonlinear strain energy terms result in pseudo generalized forces (as functions of the displacements) which are combined with the applied generalized forces. The resulting set of nonlinear algebraic equilibrium equations is solved by one of several methods: Newton-Raphson

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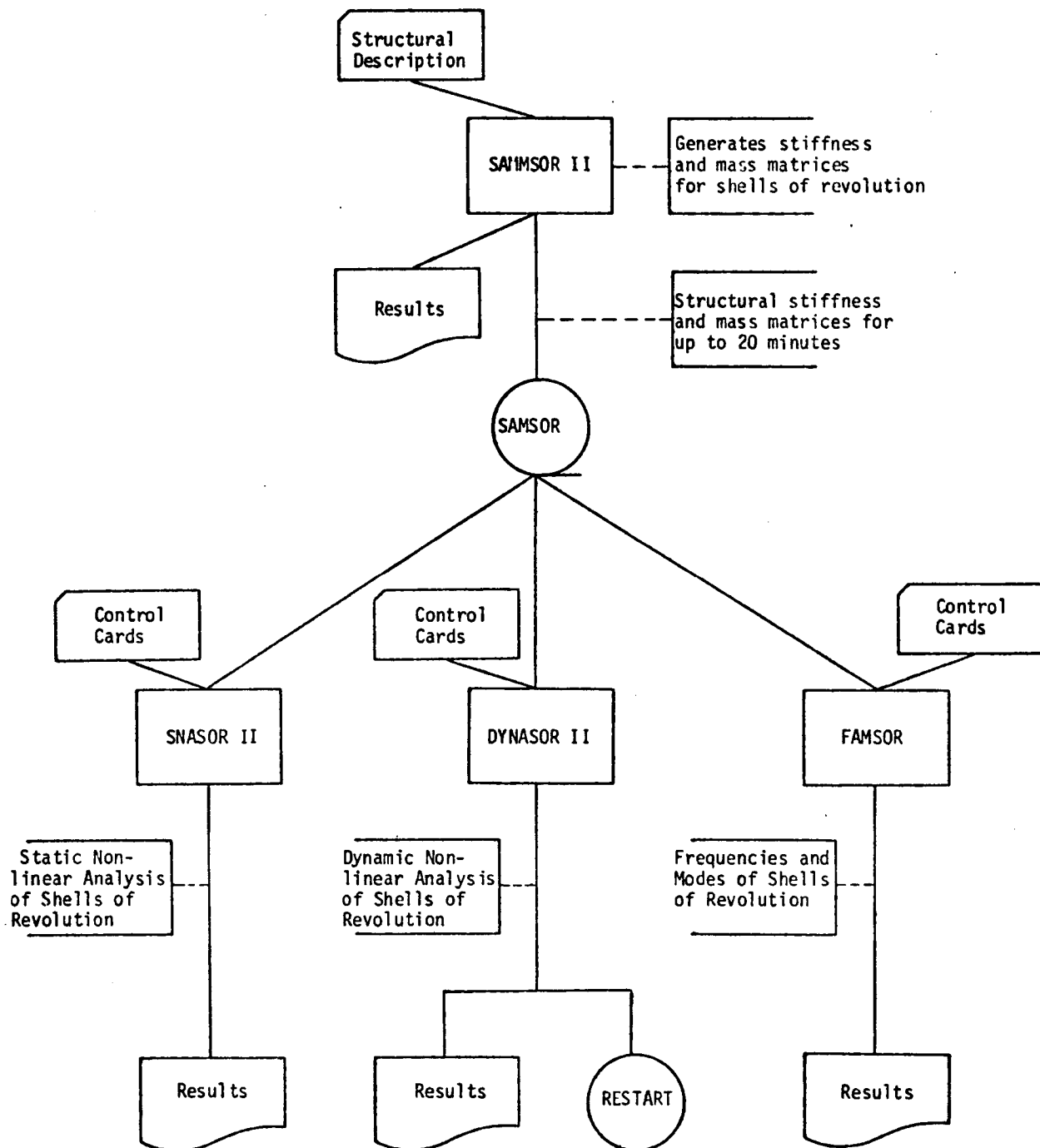
type iteration, incremental stiffness method, or a modified incremental stiffness method. In general, the Newton-Raphson procedure is the best and yields accurate results for highly nonlinear problems.

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FAMSOR - Frequencies And Modes for Shells Of Revolution can be determined using the fourth code. Using the stiffness matrix generated by SAMMSOR and a lumped mass representation developed from the consistent mass matrix generated by SAMMSOR, a specified number of natural frequencies (beginning with the lowest or fundamental frequency) are obtained using the inverse iteration method. The mode shapes for each of the frequencies are also obtained.



SYSTEM FLOWCHART



## ENVIRONMENT

The DYNASOR II program runs under OS/360 MFT or MVT and requires 220K of memory on an IBM S/360 computer. The system must also have a card reader, printer, 3 9-track tape drives and 2314 disk storage.

## SYSTEM OPERATION

## A. Run Characteristics

- 1). Functional description of run  
(see abstract)
- 2). Relationship to other runs  
(see system flowchart)
- 3). Set-up and run instructions
  - a. get SAMMSOR output tape
  - b. 2 scratch tapes
  - c. punch control cards
  - d. fill out job ticket
  - e. load job
- 4). Run frequency  
when desired
- 5). Run prerequisites  
SAMMSOR
- 6). Run control  
none
- 7). Run schedule  
no set schedule; run when desired or on demand
- 8). External procedures  
none

B.

MACHINE SET UP FORM

PROG NAME DYNASOR      PROG # \_\_\_\_\_      USER I.D. CODE \_\_\_\_\_  
 AREA Aerospace Engr.      PROJECT \_\_\_\_\_      PROGRAMMER Haisler  
 JOB PREQ SAMSOR      REGION SIZE 110K \_\_\_\_\_ K

DIRECT ACCESS REQUIREMENTS: none

PERMANENT

USER ASSIGNED PACK	DATA SET NAME	DDNAME	SERIAL # / CELL #	BIN #

TEMPORARY none

USER ASSIGNED PACK	DATA SET NAME	DDNAME	SERIAL # / CELL #	SPACE =

TAPE REQUIREMENTS:

#7 TRK. UNITS 0      #9 TRK. UNITS 3

DATA SET NAME	DDNAME	LABEL TYPE	7 OR 9 TRK	DEN/ MODE	WRITE RING IN	RET PD IN DAYS	OUTPUT FROM RUN #	INPUT TO RUN #
SAMSOR	FT01F008	SL	9		no	until next	SAMSOR	DYNASOR
none	FT01F009	SL	9		yes	run	scratch	DYNASOR
none	FT01F010	SL	9		yes		scratch	DYNASOR

(FOR ADDITIONAL INFORMATION ATTACH ANOTHER SHEET)

ADDITION       REPLACEMENT

DATE \_\_\_\_\_ SECTION \_\_\_\_\_ PAGE \_\_\_\_\_

CARD READER REQUIREMENTS: SYSIN

DATA SET NAME	DDNAME	SOURCE	DISPOSITION

CARD PUNCH REQUIREMENTS: none

DATA SET NAME	DDNAME	POCKET #	DISPOSITION

PRINTER REQUIREMENTS: SYSPRINT

DATA SET NAME	DDNAME	PRINT TRAIN	FORM #	SETUP #	LINES PER INCH	BURST	DECOL-LATE	DISPOSITION

ADDITION  REPLACEMENT

DATE \_\_\_\_\_ SECTION \_\_\_\_\_ PAGE \_\_\_\_\_

C. File Information Sheet

date of last update

name: SAMSOR

system or application: DYNASOR

description of contents and use: stiffness and mass matrices for  
up to 20 harmonies; input to DYNASOR II program

storage medium: tape

record characteristics: Fortran output, variable length records

block characteristics: 7200 byte blocks, spanned

file activity: (approximate when necessary)

- not a permanently maintained file
- created when computing shells of revolution
- restart information written on tape by DYNASOR run.

C2

## D. Job Control Language

```

//DYNASR      JØB(_____,____,....)
/*CLASS      _
//JØBLIB      DD          DSNAME=USER.DYNASØR.JØBLIB,DISP=SHR
//DYNASØR     EXEC       PGM=DYNASØR
//FT01F005    DD          DDNAME=SYSIN
//FT01F006    DD          SYSØUT=A
//FT01F007    DD          SYSØUT=B
//FT01F008    DD          UNIT=2400,DISP=(ØLD,KEEP),DSNAME=SAMSØRIN,
//              DCB=(RECFM=VBS,BLØCKSIZE=7200),VØL=SER=SAMSØ1
//FT01F009    DD          UNIT=2400,DISP=(NEW,DELETE),DCB=(RECFM=VBS,
//              BLØCKSIZE=7200),VØL=SER=DYNSR1
//FT01F010    DD          UNIT=2400,DISP=(NEW,DELETE),DCB=(RECFM=VBS,
//              BLØCKSIZE=7200),VØL=SER=DYNSR2
//SYSIN       DD          *
              .
              .
              .
              .
              .
              .
/*

```

CONTROL CARDS

## E. RUN CONTROL CARDS

Card Type I

Columns	Variable	Type
1-5	NCASES	Numeric
6-10	ND	Numeric
11-15	NS	Numeric

Card Type II-A

Columns	Variable	Type
1-5	NCARDS	Numeric
6-10	NT	Numeric

Card Type II-B

Columns	Variable	Type
1-80	COMENT	Alphanumeric

Card Type III-A

Columns	Variable	Type
1-10	TOTIME	Numeric
11-20	DELTE	Numeric
21-25	IRSTRT	Numeric
26-30	INCRST	Numeric
31-35	NCLOSE	Numeric
36-40	ITELF	Numeric



## RUN CONTROL CARDS

Card Type III-B

Columns	Variable	Type
1-5	NPRNTQ	Numeric
6-10	IPRINT	Numeric
11-15	NCLCST	Numeric
16-20	NSTRSS	Numeric
21-25	NPRNT	Numeric
26-30	NPRNIT	Numeric
31-35	NPRNTL	Numeric
36-40	NPRNTF	Numeric
41-45	NPRNTH	Numeric
46-50	NPRNMS	Numeric

Card Type IV-A

Columns	Variable	Type
1-5	NTHETA	Numeric

Card Type IV-B

Columns	Variable	Type
1-10	THETA(1)	Numeric
11-20	THETA(2)	Numeric
"	"	
"	"	
"	THETA (NTHETA)	

Card Type V

Columns	Variable	Type
1-5	NH	Numeric

## RUN CONTROL CARDS

Card Type V (Continued)

Columns	Variable	Type
6-10	IHARM(1)	Numeric
11-15	IHARM(2)	
16-20	IHARM(3)	
21-25	IHARM(4)	
26-30	IHARM(5)	

Card Type VI-A

Columns	Variable	Type
1-5	NODRES	Numeric

Card Type VI-B

Columns	Variable	Type
1-5	NP	Numeric
6-10	NDIRCT	Numeric

Card Type VII-A

Columns	Variable	Type
1-5	IQN	Numeric
6-10	IQN1	Numeric

Card Type VII-B

Columns	Variable	Type
1-5	IN1	Numeric
6-10	IN2	Numeric

## RUN CONTROL CARDS

Card Type VII-B (Continued)

Columns	Variable	Type
11-20	$\dot{q}_1$	Numeric
21-30	$\dot{q}_2$	Numeric
31-40	$\dot{q}_3$	Numeric
41-50	$\dot{q}_4$	Numeric

Card Type VII-C

Columns	Variable	Type
1-5	IN1	Numeric
6-10	IN2	Numeric
11-20	$\dot{q}_1$	Numeric
21-30	$\dot{q}_2$	Numeric
31-40	$\dot{q}_3$	Numeric
41-50	$\dot{q}_4$	Numeric

Card Type VIII

Columns	Variable	Type
1-5	IELM1	Numeric
6-10	IELM2	Numeric
11-20	ALSI1	Numeric
21-30	ALTI1	Numeric

Card Type IX-A

Columns	Variable	Type
1-10	T1	Numeric

## RUN CONTROL CARDS

Card Type IX-A (Continued)

Columns	Variable	Type
11-15	NCF	Numeric
16-20	IDELF	Numeric
21-25	IDCOE	Numeric
26-30	ITCOE	Numeric
31-38	CONSTF	Alphanumeric

Card Type IX-B-1

Columns	Variable	Type
1-5	NCF1	Numeric

Card Type IX-B-2

Columns	Variable	Type
1-5	IN1	Numeric
6-10	IN2	Numeric
11-20	F1	Numeric
21-30	F2	Numeric
31-40	F3	Numeric
41-50	F4	Numeric

Card Type IX-C-1-a

Columns	Variable	Type
1-5	IELM1	Numeric
6-10	IELM2	Numeric
11-15	NDP	Numeric

## RUN CONTROL CARDS

Card Type IX-C-1-b

Columns	Variable	Type
1-10	THETB	Numeric
11-20	P	Numeric
21-30	R	Numeric
31-40	S	Numeric

Card Type IX-C-2-a

Columns	Variable	Type
1-5	IELM1	Numeric
6-10	IELM2	Numeric

Card Type IX-C-2-b

Columns	Variable	Type
1-10	P	Numeric
11-20	R	Numeric
21-30	S	Numeric

Card Type IX-D-1-a

Columns	Variable	Type
1-5	IELM1	Numeric
6-10	IELM2	Numeric
11-15	NDP	Numeric

Card Type IX-D-1-b

Columns	Variable	Type
1-10	THETB	Numeric
11-20	P	Numeric
21-30	R	Numeric

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RUN CONTROL CARDS

Card Type IX-D-2-a

Columns	Variable	Type
1-5	IELM1	Numeric
6-10	IELM2	Numeric

Card Type IX-D-2-b

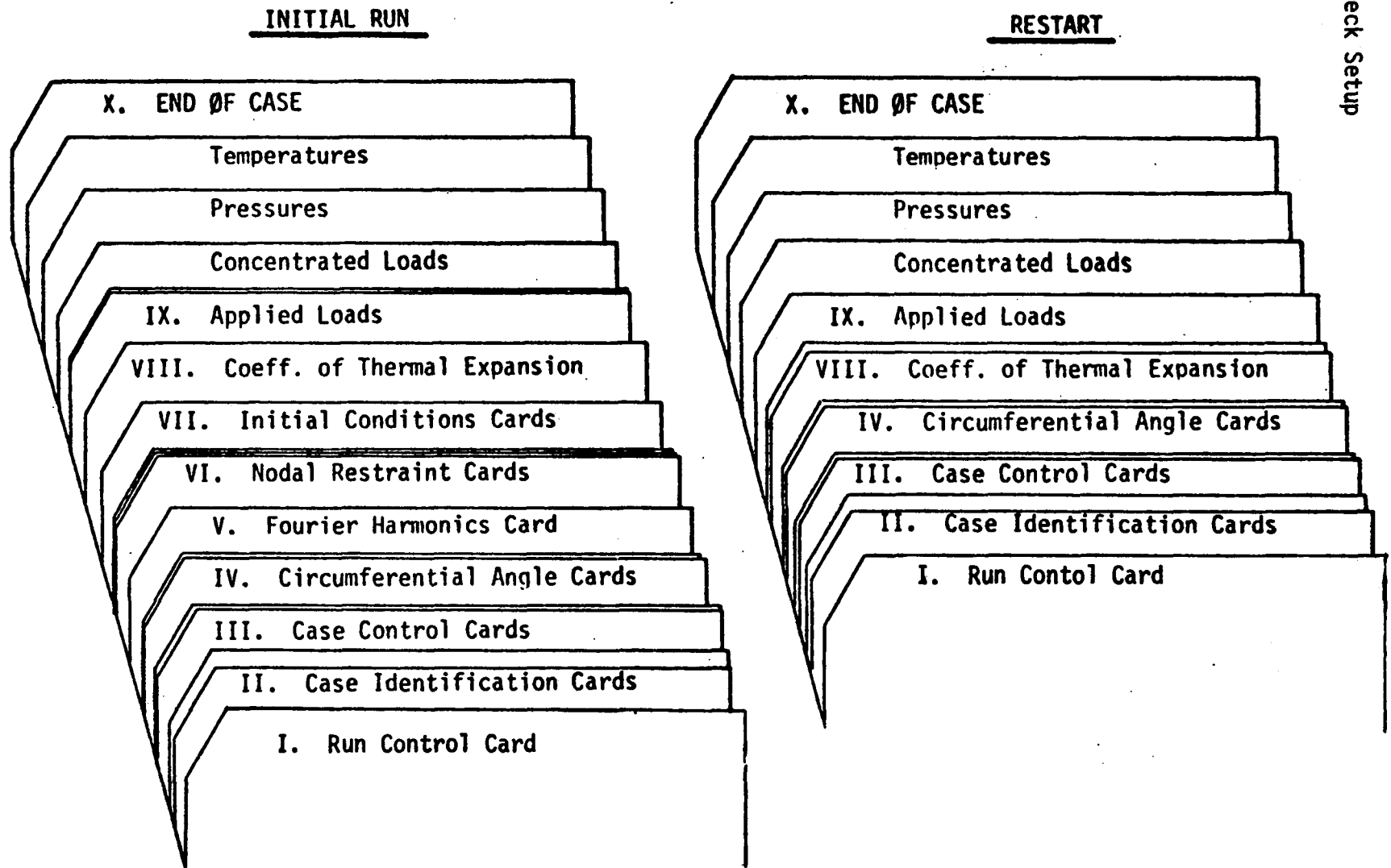
Columns	Variable	Type
1-10	TH1	Numeric
11-20	DTH1	Numeric

Card Type X

Columns	Punch
1-11	END OF CASE

Card Type XI

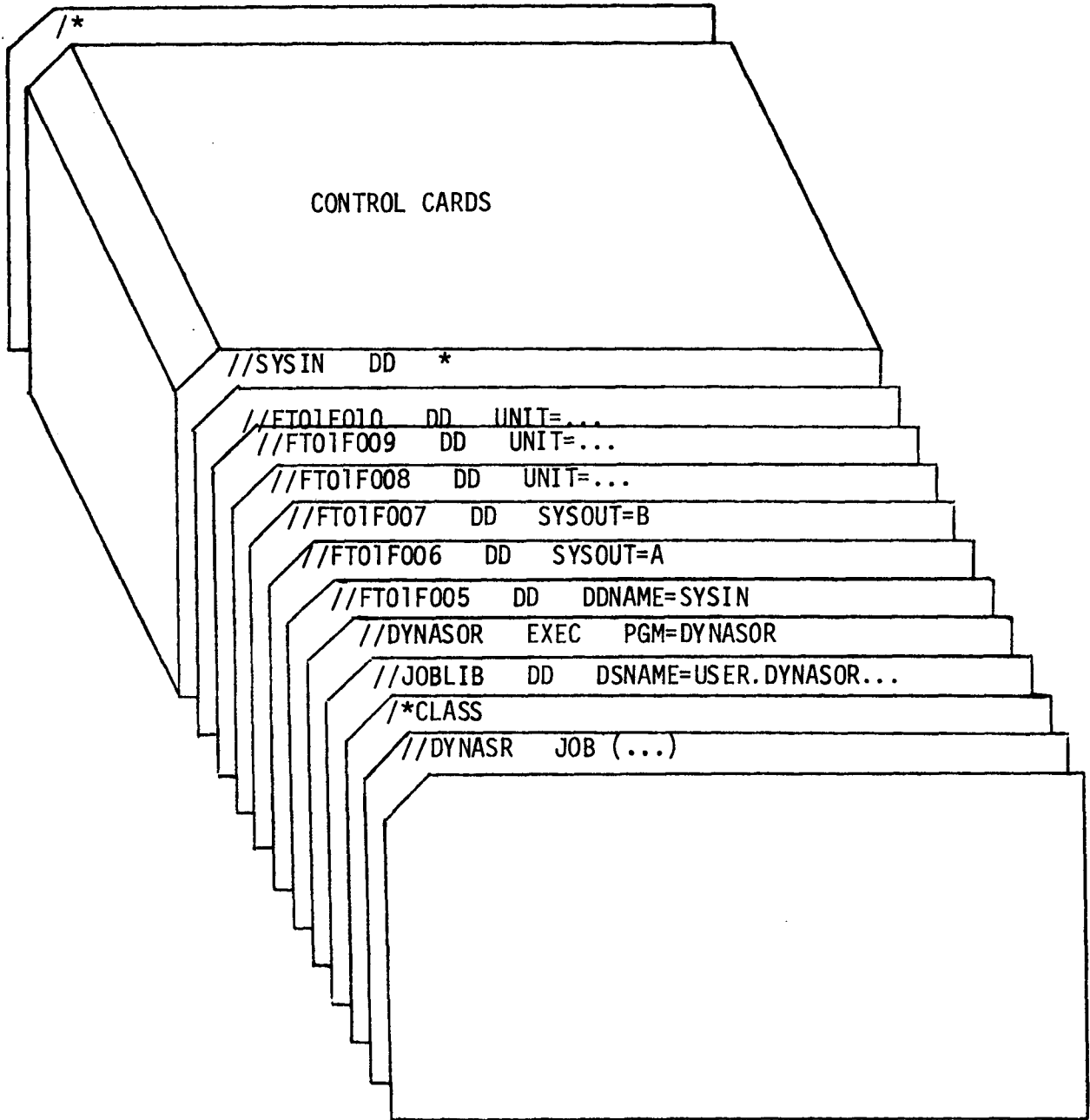
Columns	Punch
1-10	END OF RUN



F. Deck Setup

FIG. 1 CONSTITUTION OF DATA DECKS - INITIAL RUN AND RESTART MODES.

DECK SETUP FOR DYNASOR II





G. 360 MESSAGE FORM

DATE \_\_\_\_\_

ADDITION  REPLACEMENT

OPERATIONAL

ADMINISTRATIVE

Jobname \_\_\_\_\_ PROGRAMMER \_\_\_\_\_ MSG ID \_\_\_\_\_

MESSAGE: The number of input cases does not agree with the value of  
ncases input

MEANING:

ACTION:

360 MESSAGE FORM

DATE \_\_\_\_\_

ADDITION  REPLACEMENT

OPERATIONAL

ADMINISTRATIVE

Jobname \_\_\_\_\_ PROGRAMMER \_\_\_\_\_ MSG ID \_\_\_\_\_

MESSAGE: Restart information for time increment no., I5,/,10X, corresponding to time, F12.4, microseconds,/,2X, has been placed on tape for use in subsequent runs//

MEANING:

ACTION:

360 MESSAGE FORM

DATE \_\_\_\_\_

ADDITION  REPLACEMENT

OPERATIONAL

ADMINISTRATIVE

Jobname \_\_\_\_\_ PROGRAMMER \_\_\_\_\_ MSG ID \_\_\_\_\_

MESSAGE: ITAM, I5, 5X time, E 12.5 Execution terminated - displacements  
greater than 1.E + 4

MEANING:

ACTION:

H. CHECKPOINT, RESTART, ERROR PROCEDURES, BACKUP, AND RECOVERY PROCEDURES

Recovery Procedures:

If many computations are to be done, a backup copy of the SAMMSOR input and restart tape can be made. Otherwise, no backup is required.

Restart Procedures:

To restart DYNASOR program, set IRSTRT = 1 on control card type III-A. Use SAMMSOR input tape. Restart information was written onto this tape in the previous run.

DYNASOR II - A FINITE ELEMENT PROGRAM FOR THE  
DYNAMIC NONLINEAR ANALYSIS OF SHELLS OF REVOLUTION

Joe R. Tillerson and Walter E. Haisler

MAINTENANCE MANUAL

October 15, 1970

TEES-RPT-70-19

Texas A&M University  
College Station, Texas

J-3-1

## ABSTRACT

The DYNASOR II program is used for the Dynamic Nonlinear Analysis of Shells Of Revolution. The equations of motion of the shell are solved using Houbolt's numerical procedure. The displacements and stress resultants can be determined for both symmetrical and asymmetrical loading conditions. Asymmetrical dynamic buckling can be investigated. Solutions can be obtained for highly nonlinear problems utilizing as many as five of the harmonics generated by SAMMSOR program. A restart capability allows the user to restart the program at a specified time.

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## SYSTEM OVERVIEW

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## SOR - Shell Of Revolution

Preceding page blank

## Computer Programs

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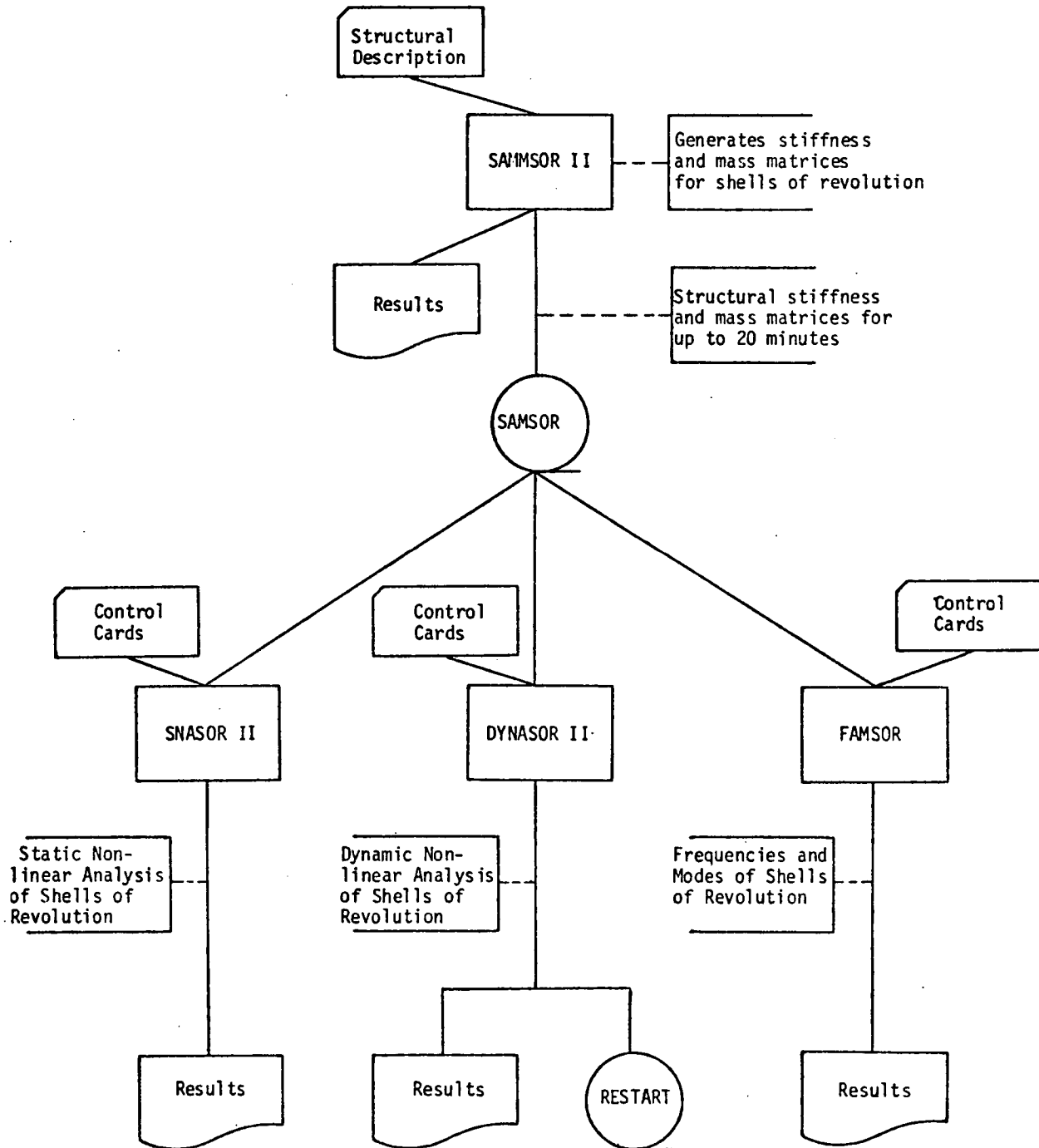
October 1972

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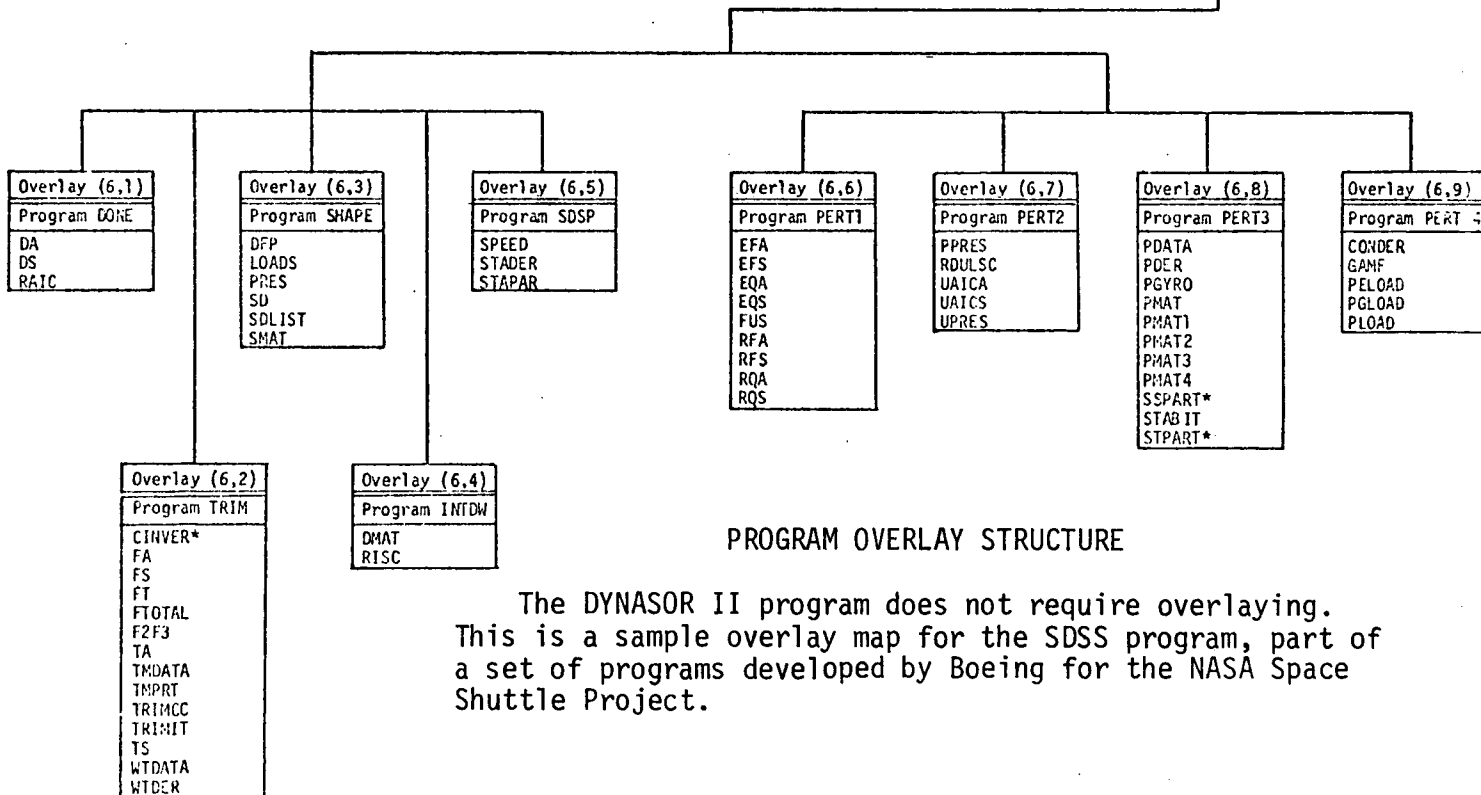
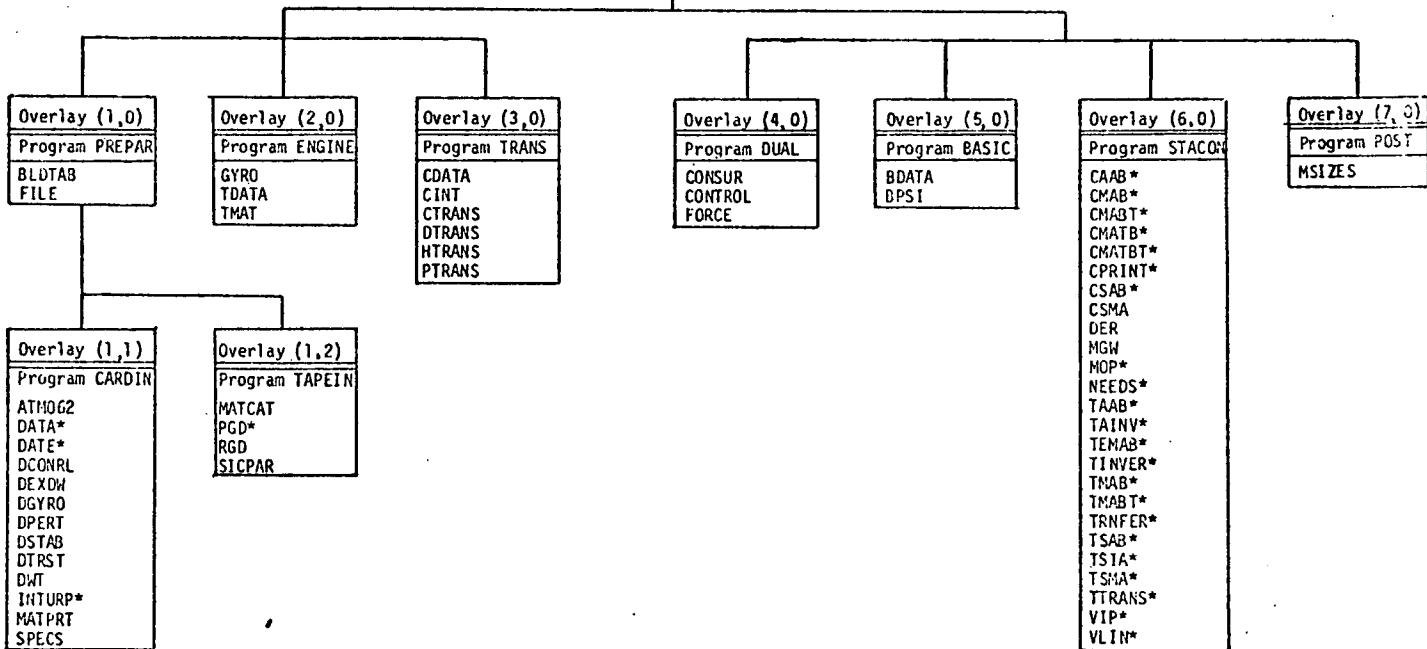
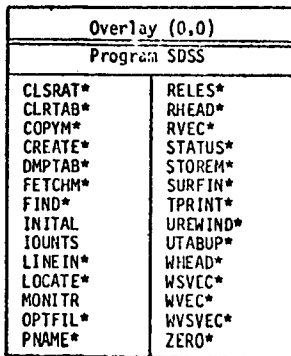
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SYSTEM FLOWCHART



## ENVIRONMENT

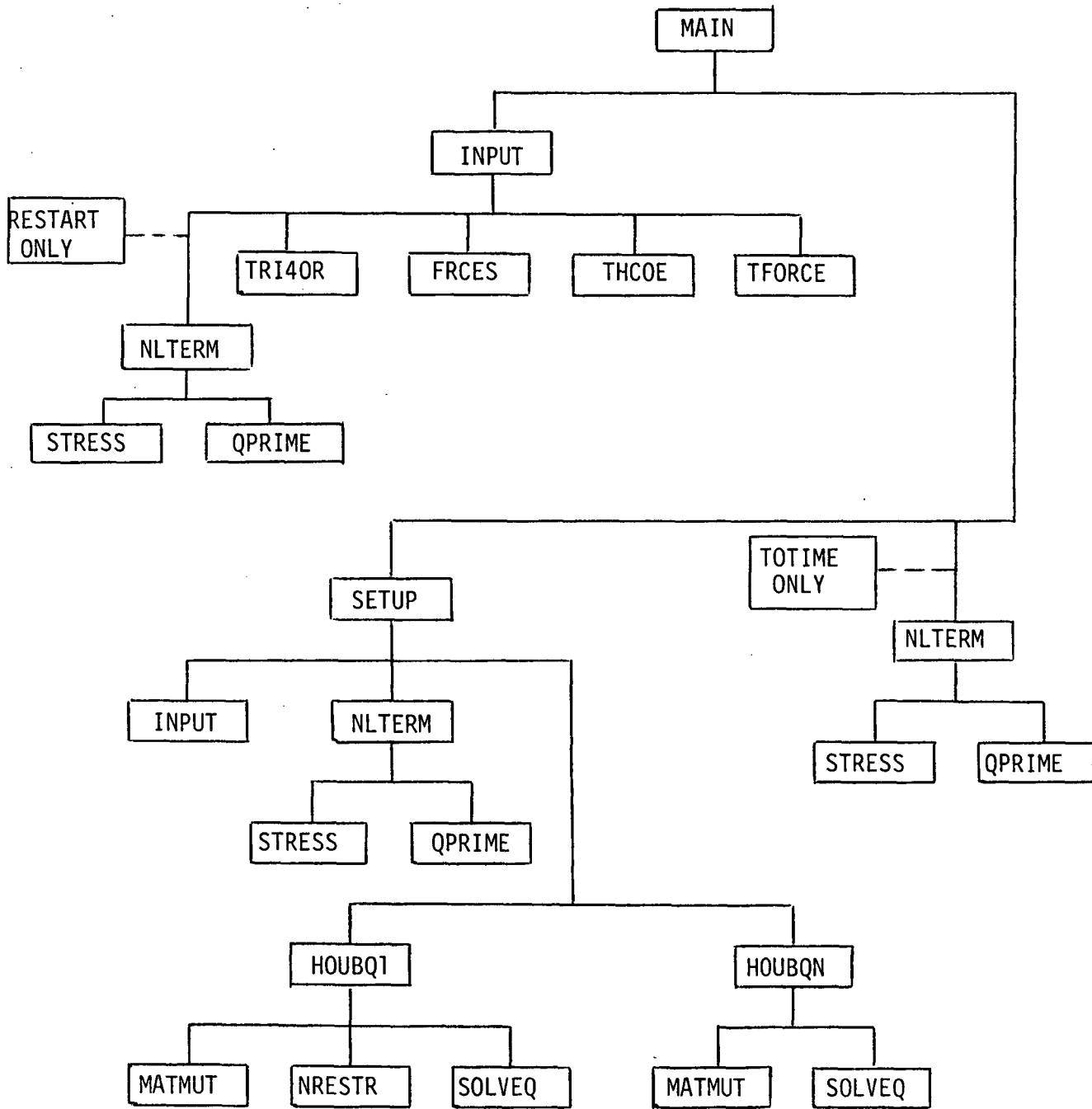
The DYNASOR II program runs under OS/360 MFT or MVT and requires 220K of memory on an IBM S/360 computer. The system must also have a card reader, printer, 3 9-track tape drives and 2314 disk storage.



PROGRAM OVERLAY STRUCTURE

The DYNASOR II program does not require overlaying. This is a sample overlay map for the SDSS program, part of a set of programs developed by Boeing for the NASA Space Shuttle Project.





SUBROUTINE CALL DIAGRAM

```

CE(FRCES) DYN14872
C DYN14874
C DESCRIPTION - TO READ IN THE DISTRIBUTED LOADS ON THE SHELL DYN14876
C STRUCTURE. THEN, USING EITHER CIRCUMFERENTIAL OR DYN14878
C FOURIER COEFFICIENTS DATA, CALCULATE THE LINEAR DYN14880
C GENERALIZED LOADS. DYN14882
C DYN14884
C INPUT ARGUMENTS. DYN14886
C IB = FORCE ARRAY STEPPING PARAMETER, USED TO MODIFY CURRENT DYN14888
C BLOCK OF STORAGE FOR FORCE. DYN14890
C IELM = NUMBER OF SHELL ELEMENTS. DYN14892
C DYN14894
C OUTPUT ARGUMENTS. DYN14896
C FORCE = MATRIX OF GENERALIZED FORCES DUE TO EXTERNAL LOADS AND DYN14898
C TEMPERATURES. DYN14900
C Q THERMAL COEFFICIENTS USED IN CALCULATING GENERALIZED DYN14902
C LINEAR LOADS DUE TO THERMAL EFFECTS. DYN14904
C QQ = GENERALIZED LINEAR LOADS DUE TO THERMAL EFFECTS. DYN14906
C DYN14908
C EXTERNALS. DYN14910
C CALLED BY DYN14912
C INPUT DYN14914
C DYN14916
0001 SUBROUTINE FRCES (IELM,ALPHK,IB) DYN14918
0002 IMPLICIT REAL*8 (A-H,O-Z) DYN14920
0003 COMMON /FRCE/ P(74),R(74),S(74),THETB(74) DYN14922
0004 COMMON /CHALS/ AL(167),CHECK(8,8) DYN14924
0005 COMMON /QS/ QN(1020),QN1(1020),FORCE(2040),QP(1020),QP1(1020), DYN14926
1 QN2(1020) DYN14928
0006 COMMON /CONST/ NH,NELEMS,NNODES,NSIZE,NPRNTQ,NEQ,NEQT,N,NN,NHNS, DYN14930
1 DT2,NPRNTL,NPRNTF,IDELEF,IDCOE DYN14940
0007 COMMON /HARM/ NHP,IHARM(5) DYN14950
0008 COMMON /QUES/ Q(8),QQ(8) DYN14960
0009 COMMON /TAPES/ NT,ND,NS DYN14970
C1 READ DISTRIBUTED LOADS USING CIRCUMFERENTIAL ANGLES OR FOURIER DYN14972
C1C COEFFICIENTS DYN14974
0010 PI=3.14159265 DYN15020
0011 IF (IELM.EQ.1) IELM2=0 DYN15030
0012 IF (IELM.LE.IELM2.AND.IELM.NE.1) GO TO 40 DYN15040
0013 IF (IDCOE.NE.1) GO TO 10 DYN15050
0014 IF (NPRNTL.EQ.1.AND.IELM.EQ.1) WRITE (6,160) DYN15060
C READ INPUT DATA FOR CARD TYPE IX - C - 2 DYN15070
0015 READ (ND,170) IELM1,IELM2,(P(I),R(I),S(I),I=1,NH) DYN15080
0016 IF (NPRNTL.EQ.1) WRITE (6,180) IELM1,IELM2,(P(I),R(I),S(I),I=1, DYN15090
1 NH) DYN15100
0017 GO TO 40 DYN15110
0018 10 CONTINUE DYN15120
0019 IF (NPRNTL.EQ.1.AND.IELM.EQ.1) WRITE (6,190) DYN15130

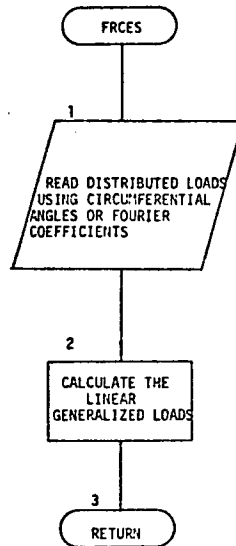
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	C	READ DISTRIBUTED LOADS	DYN15140
	C		DYN15150
	C	READ INPUT DATA FOR CARD TYPE IX - C - 1	DYN15160
0020		READ (ND,200) IELM1,IELM2,NDP,(THETB(I),P(I),R(I),S(I),I=1,NDP)	DYN15170
0021		NDP2=2*NDP+1	DYN15180
	C		DYN15188
0022		DO 20 IF=1,NDP	DYN15190
0023		ANG=360.0-THETB(IF)	DYN15200
0024		KEY=NDP2-IF	DYN15210
0025		THETB(KEY)=ANG	DYN15220
0026		P(KEY)=P(IF)	DYN15230
0027		R(KEY)=R(IF)	DYN15240
0028		S(KEY)=S(IF)	DYN15250
0029		20 CONTINUE	DYN15260
	C		DYN15263
0030		IF (NPRNTL.EQ.1) WRITE (6,210) IELM1,IELM2,(P(I),R(I),S(I),	DYN15270
	1	THETB(I),THETB(I+1),I=1,NDP)	DYN15280
0031		ND2=2*NDP	DYN15290
0032		NDPP2=NDP+2	DYN15300
0033		IF (NPRNTL.EQ.1.AND.NDP.GT.1) WRITE (6,220) (P(I),R(I),S(I),	DYN15310
	1	THETB(I-1),THETB(I),I=NDPP2,ND2)	DYN15320
	C		DYN15328
0034		DO 30 IDP=1,ND2	DYN15330
0035		THETB(IDP)=THETB(IDP)/57.2957795	DYN15340
0036		30 CONTINUE	DYN15342
	C		DYN15345
	C1	CALCULATE THE LINEAR GENERALIZED LOADS	DYN15347
0037		40 CONTINUE	DYN15350
	C		DYN15358
0038		DO 150 IH=1,NH	DYN15360
0039		KYP=IHARM(IH)	DYN15370
0040		YKP=KYP	DYN15380
0041		IF (IDCOE.EQ.1) GO TO 110	DYN15390
	C		DYN15398
0042		DO 50 I=1,8	DYN15400
0043		Q(I)=0.0	DYN15410
0044		50 CONTINUE	DYN15412
	C		DYN15415
0045		NDP1=2.0*NDP-1	DYN15420
	C		DYN15428
0046		DO 100 I=1,NDP1	DYN15430
0047		IF (NDP.EQ.1) GO TO 70	DYN15440
0048		X1=THETB(I)*YKP	DYN15450
0049		X2=THETB(I+1)*YKP	DYN15460
0050		IF (KYP.GT.0) GO TO 60	DYN15470
0051		PINT=P(I)*(THETB(I+1)-THETB(I))	DYN15480
0052		RINT=R(I)*(THETB(I+1)-THETB(I))	DYN15490
0053		SINT=0.0	DYN15500

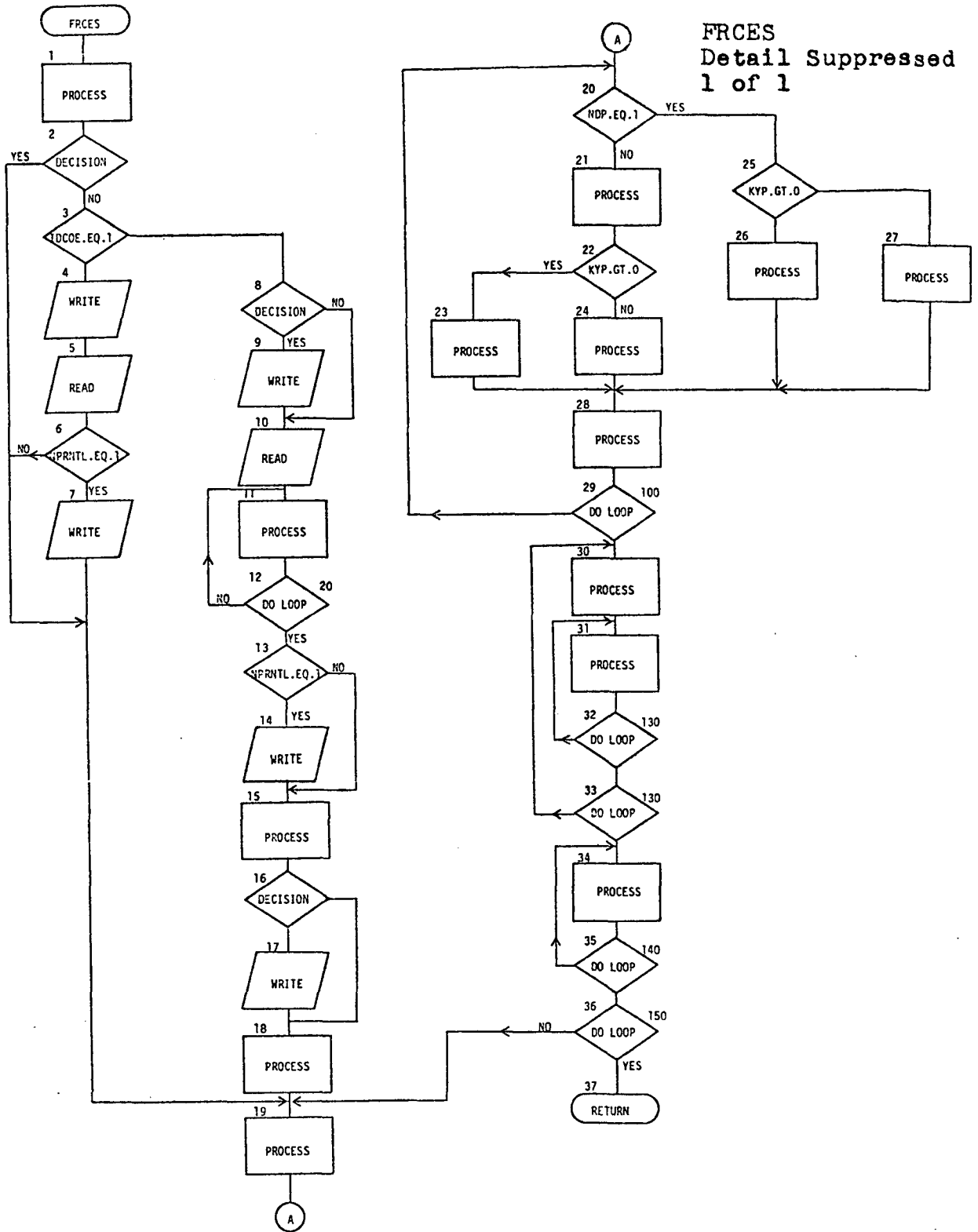
0054		GO TO 90	DYN15510
0055	60	PINT=P(I)*(DSIN(X2)-DSIN(X1))/YKP	DYN15520
0056		RINT=R(I)*(DSIN(X2)-DSIN(X1))/YKP	DYN15530
0057		SINT=-S(I)*(DCOS(X2)-DCOS(X1))/YKP	DYN15540
0058		GO TO 90	DYN15550
0059	70	IF (KYP.GT.0) GO TO 80	DYN15560
0060		PINT=2.0*PI*P(1)	DYN15570
0061		RINT=2.0*PI*R(1)	DYN15580
0062		SINT=C.0	DYN15590
0063		GO TO 90	DYN15600
0064	80	PINT=0.0	DYN15610
0065		RINT=0.0	DYN15620
0066		SINT=0.0	DYN15630
0067	90	Q(1)=Q(1)+RINT*AL(1)	DYN15640
0068		Q(2)=Q(2)+RINT*AL(2)	DYN15650
0069		Q(3)=Q(3)+RINT*AL(3)	DYN15660
0070		Q(4)=Q(4)+RINT*AL(157)	DYN15670
0071		Q(5)=Q(5)+PINT*AL(1)	DYN15680
0072		Q(6)=Q(6)+PINT*AL(2)	DYN15690
0073		Q(7)=Q(7)+SINT*AL(1)	DYN15700
0074		Q(8)=Q(8)+SINT*AL(2)	DYN15710
0075	100	CONTINUE	DYN15712
	C		DYN15715
0076		GO TO 120	DYN15720
0077	110	CONTINUE	DYN15730
0078		C12=1.0	DYN15740
0079		IF (KYP.EQ.0) C12=2.0	DYN15750
0080		PINT=C12*PI*P(IH)	DYN15760
0081		RINT=C12*PI*R(IH)	DYN15770
0082		SINT=(2.0-C12)*PI*S(IH)	DYN15780
0083		Q(1)=RINT*AL(1)	DYN15790
0084		Q(2)=RINT*AL(2)	DYN15800
0085		Q(3)=RINT*AL(3)	DYN15810
0086		Q(4)=RINT*AL(157)	DYN15820
0087		Q(5)=PINT*AL(1)	DYN15830
0088		Q(6)=PINT*AL(2)	DYN15840
0089		Q(7)=SINT*AL(1)	DYN15850
0090		Q(8)=SINT*AL(2)	DYN15860
0091	120	CONTINUE	DYN15870
	C		DYN15878
0092		DO 130 I=1,8	DYN15880
0093		QQ(I)=0.0	DYN15890
	C		DYN15898
0094		DO 130 J=1,8	DYN15900
0095		QQ(I)=QQ(I)+CHECK(J,I)*Q(J)	DYN15910
0096	130	CONTINUE	DYN15912
	C		DYN15915
	C		DYN15918

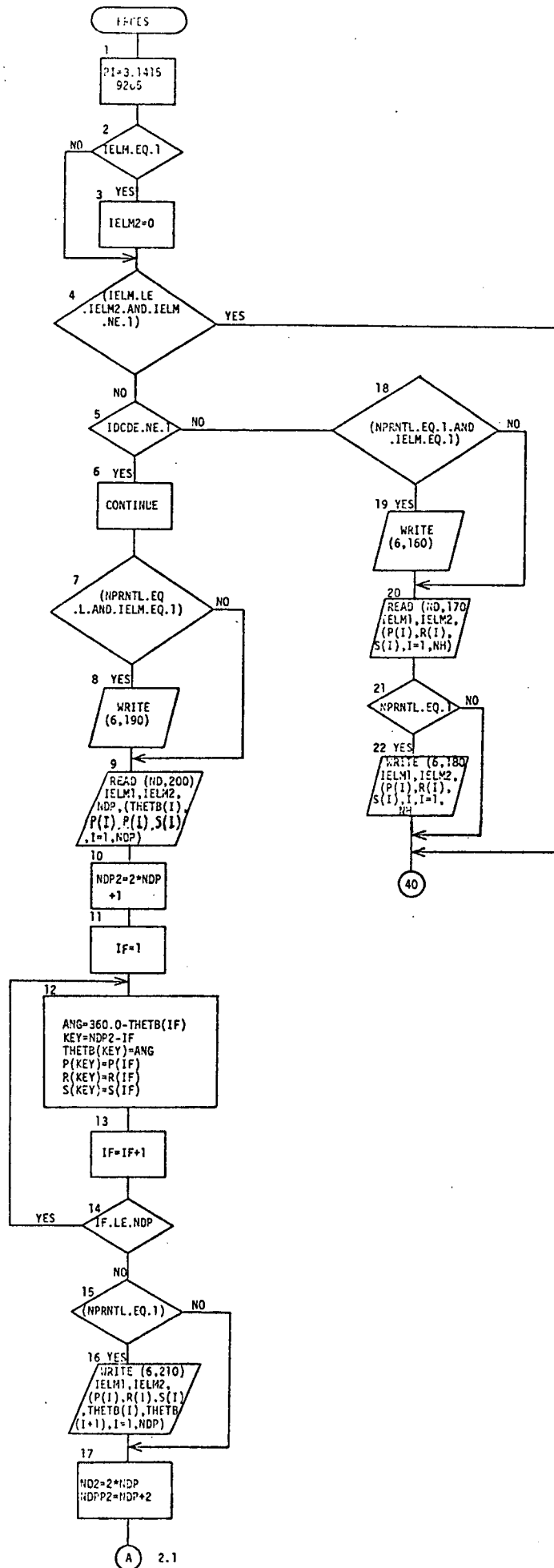
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0097          DO 140 I=1,8                      DYN15920
0098             J=4*(IELM-1)+I+(IH-1)*NEQ+IB*NEQT DYN15930
0099             FORCE(J)=FORCE(J)+QQ(I)         DYN15940
0100          140 CONTINUE                      DYN15942
           C                                  DYN15945
0101          150 CONTINUE                      DYN15950
           C                                  DYN15953
0102          RETURN                            DYN15960
           C                                  DYN15970
0103          160 FORMAT (1H1,35X,40HFOURIER COEFFICIENTS OF APPLIED PRESSURE,
           1             9H LOADINGS//          DYN15980
           1             20X,10HMERIDIANAL,20X,6HNORMAL,20X,10HTANGENTIAL,10X, DYN15990
           2             12HHARMONIC NO.//)     DYN16000
0104          170 FORMAT (2I5/(3F10.0))        DYN16010
0105          180 FORMAT (/60X,11HELEMENT NO.,I3,1H-,I3,/(2X,3D28.7,15X,I2)) DYN16020
0106          190 FORMAT (1H1,51X,30HAPPLIED LOADS ON THE STRUCTURE///
           1             56X,19HPRESSURE COMPONENTS// DYN16032
           2             20X,10HMERIDIANAL,20X,6HNORMAL,20X,10HTANGENTIAL,11X, DYN16040
           2             19HFROM THETA TO THETA,9H(DEGREES)) DYN16050
0107          200 FORMAT (3I5/(4F10.0))        DYN16060
0108          210 FORMAT (/60X,11HELEMENT NO.,I3,1H-,I2/(2X,3F28.3,12X,2F10.3)) DYN16070
0109          220 FORMAT (2X,3F28.3,12X,2F10.3) DYN16080
0110          END                              DYN16090
    
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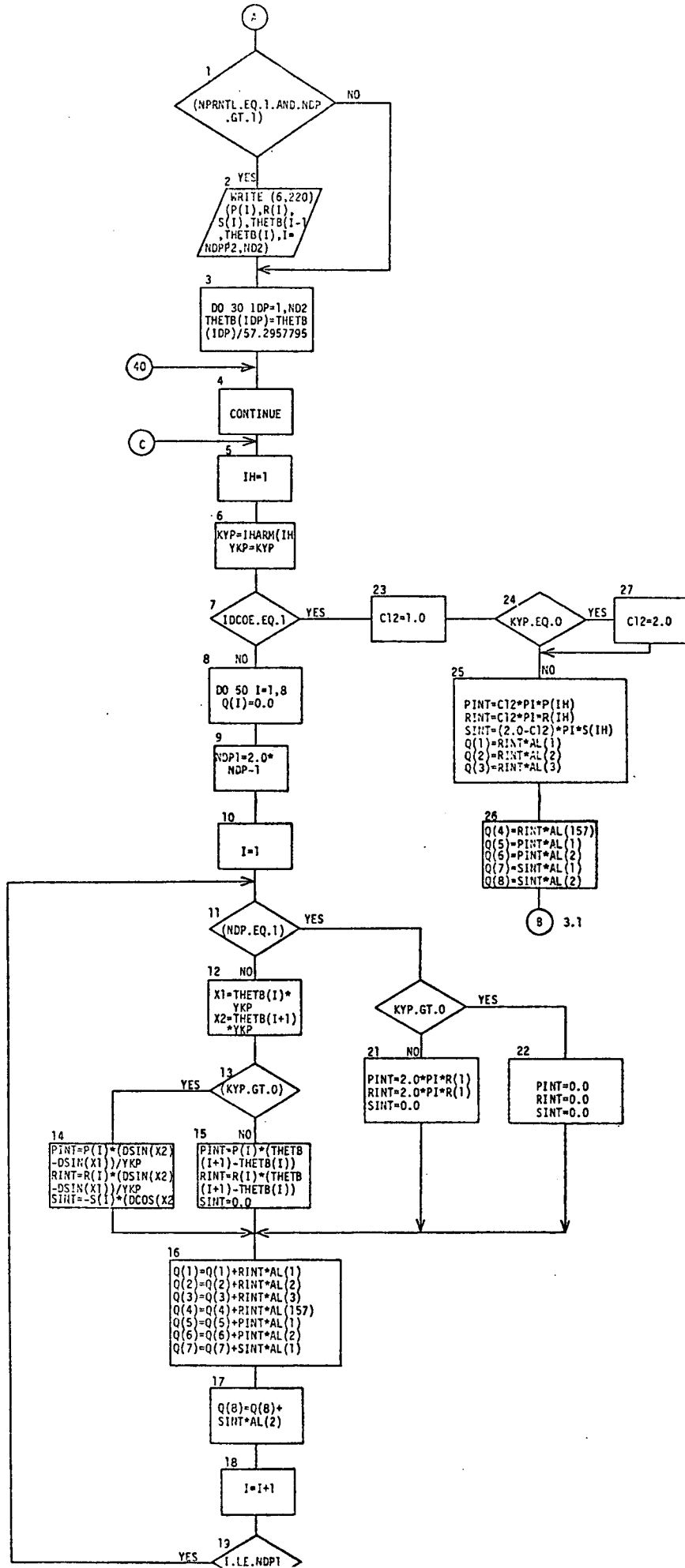


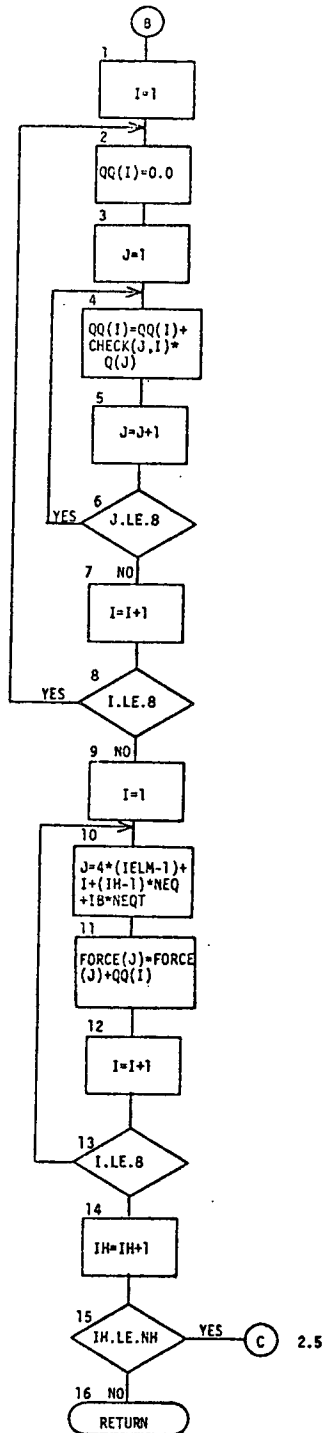
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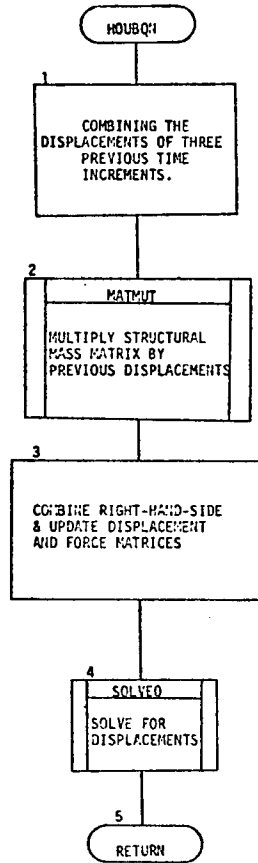
C(HOUBQN) DYNC9772
C DYNC9774
C DESCRIPTION - TO SET UP AND CONTROL THE SOLUTION OF THE DYN09776
C EQUATIONS OF MOTION FOR EACH TIME STEP EXCEPT THE FIRST ONE. DYN09778
C DYN09780
C INPUT ARGUMENTS. DYN09782
C FORCE = MATRIX OF GENERALIZED FORCES DUE TO EXTERNAL LOADS AND DYN09784
C TEMPERATURES. DYN09786
C IH = HARMONIC KEY. DYN09788
C KY = RESTART KEY. DYN09790
C QN = DISPLACEMENTS AT TIME INCREMENT (N-1) UP TO STATEMENT 20 DYN09792
C AFTER STATEMENT 30 THIS MATRIX HAS BEEN CHANGED TO DYN09794
C THE DISPLACEMENTS AT TIME STEP (N). DYN09796
C QN1 = DISPLACEMENTS AT TIME INCREMENT (N-2) BEFORE STATEMENT 10 DYN09798
C AND AT TIME INCREMENT (N-1) AFTER STATEMENT 20. DYN09800
C QP = Q - PARTIAL DERIVATIVE OF U-NL WITH RESPECT TO LOWER DYN09802
C CASE Q AT TIME STEP (N-1). DYN09804
C QP1 = Q - PARTIAL DERIVATIVE OF U-NL WITH RESPECT TO LOWER CASE DYN09806
C Q AT TIME STEP (N-2). DYN09808
C DYN09810
C OUTPUT ARGUMENTS. DYN09812
C QLOAD = RIGHT-HAND SIDE OF THE DYNAMIC EQUATIONS OF MOTION DYN09814
C BEFORE CALLING SOLVEQ. DYN09816
C QN = DISPLACEMENTS AT TIME INCREMENT (N-1) UP TO STATEMENT 20 DYN09818
C AFTER STATEMENT 30 THIS MATRIX HAS BEEN CHANGED TO DYN09820
C THE DISPLACEMENTS AT TIME STEP (N). DYN09822
C QN1 = DISPLACEMENTS AT TIME INCREMENT (N-2) BEFORE STATEMENT 10 DYN09824
C AND AT TIME INCREMENT (N-1) AFTER STATEMENT 20. DYN09826
C QN2 = DISPLACEMENTS AT TIME INCREMENT (N-3) BEFORE STATEMENT 10 DYN09828
C AND AT TIME STEP (N-2) AFTER STATEMENT 20. DYN09830
C QP1 = Q - PARTIAL DERIVATIVE OF U-NL WITH RESPECT TO LOWER CASE DYN09832
C Q AT TIME STEP (N-2). DYN09834
C DYN09836
C EXTERNALS. DYN09838
C CALLED BY DYN09840
C SETUP DYN09842
C CALLS DYN09844
C MATMUT DYN09846
C SOLVEQ DYN09848
C DYN09850
C SUBROUTINE HOUBQN (KY,IH) DYN09852
C IMPLICIT REAL*8 (A-H,O-Z) DYN09854
C COMMON /SLVEEQ/ XN(6550),QLOAD(204) DYN09856
C COMMON /QS/ QN(1020),QN1(1020),FORCE(2040),QP(1020),QP1(1020), DYN09858
C QN2(1020) DYN09860
C COMMON /RSTRNT/ NODRES,NCLOSE,LK(204) DYN09862
C COMMON /CONST/ NH,NELEMS,NNODES,NSIZE,NPRNTQ,NEQ,NEQT,N,NN,NHNS, DYN09864
C DT2,NPRNTL,NPRNTF,IDELF,IDCOE DYN09866
0001
0002
0003
0004
0005
0006

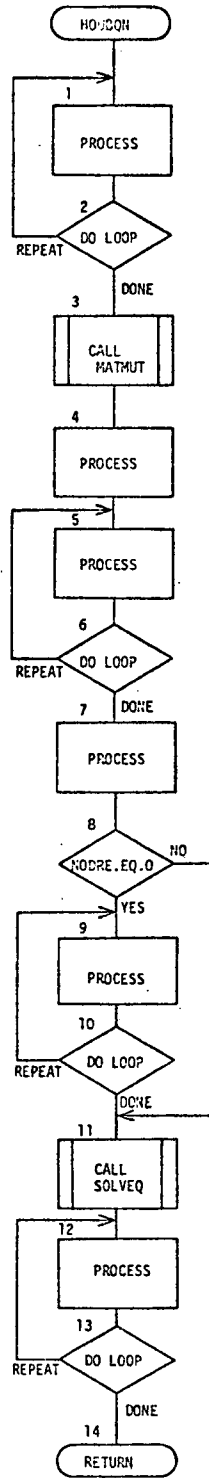
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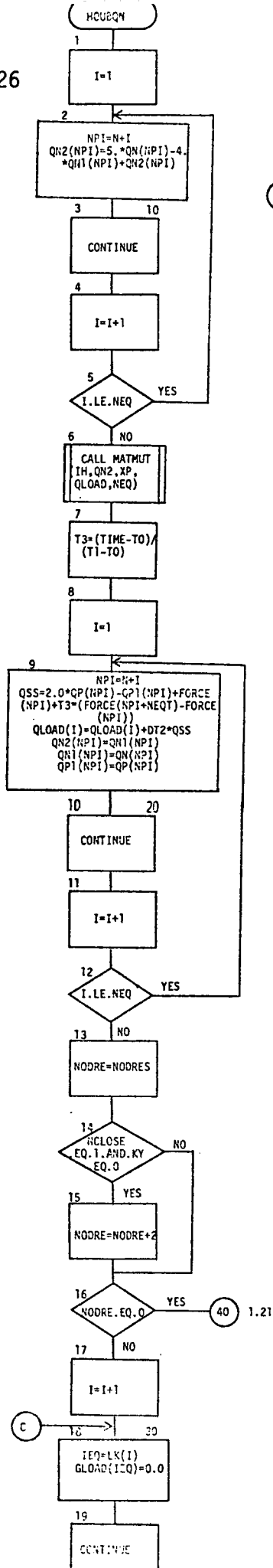
0007      COMMON /PS/ XP(6550)                                DYN09868
0008      COMMON /TMFT/ TOTIME,DELTE,TIME,TO,T1              DYN09870
      C1  COMBINING THE DISPLACEMENTS OF THREE PREVIOUS TIME INCREMENTS DYN09872
      C
0009      DO 10 I=1,NEQ                                       DYN09918
0010          NPI=N+I                                         DYN09920
0011          QN2(NPI)=5.*QN(NPI)-4.*QN1(NPI)+QN2(NPI)      DYN09930
0012      10 CONTINUE                                         DYN09940
      C
      C1  MULTIPLY STRUCTURAL MASS MATRIX BY PREVIOUS DISPLACEMENTS DYN09950
      CALL MATMUT (IH,QN2,XP,QLOAD,NEQ)                     DYN09953
0013      C1  COMBINE RIGHT-HAND-SIDE & UPDATE DISPLACEMENT AND FORCE MATRICES DYN09955
      T3=(TIME-TO)/(T1-TO)                                  DYN09960
0014      C1  CALL MATMUT (IH,QN2,XP,QLOAD,NEQ)             DYN09962
      T3=(TIME-TO)/(T1-TO)                                  DYN09980
      C
0015      DO 20 I=1,NEQ                                       DYN09988
0016          NPI=N+I                                         DYN09990
0017          QSS=2.0*QP(NPI)-QP1(NPI)+FORCE(NPI)+T3*(FORCE(NPI+NEQT)- DYN10000
      1          FORCE(NPI))                                   DYN10010
0018          QLOAD(I)=QLOAD(I)+DT2*QSS                       DYN10012
0019          QN2(NPI)=QN1(NPI)                               DYN10030
0020          QN1(NPI)=QN(NPI)                               DYN10040
0021          QP1(NPI)=QP(NPI)                               DYN10050
0022      20 CONTINUE                                         DYN10060
      C
      C1  APPLY BOUNDARY CONDITIONS TO RIGHT-HAND-SIDE      DYN10070
      NODRE=NODRES                                          DYN10073
0023      IF (NCLOSE.EQ.1.AND.KY.EQ.0) NODRE=NODRE+2      DYN10075
0024      IF (NODRE.EQ.0) GO TO 40                          DYN10090
0025      C
      C1  DO 30 I=1,NODRE                                     DYN10100
0026          IEQ=LK(I)                                       DYN10110
0027          QLOAD(IEQ)=0.0                                  DYN10118
0028      30 CONTINUE                                         DYN10120
      C
0029      C1  40 CONTINUE                                     DYN10130
      SOLVE FOR DISPLACEMENTS                              DYN10140
0030      CALL SOLVEQ (IH)                                    DYN10150
      C
0031      DO 50 I=1,NEQ                                       DYN10153
0032          NPI=N+I                                         DYN10160
0033          QN(NPI)=QLOAD(I)                               DYN10162
0034      50 CONTINUE                                         DYN10180
      C
0035      RETURN                                             DYN10190
0036      END                                               DYN10200
0037

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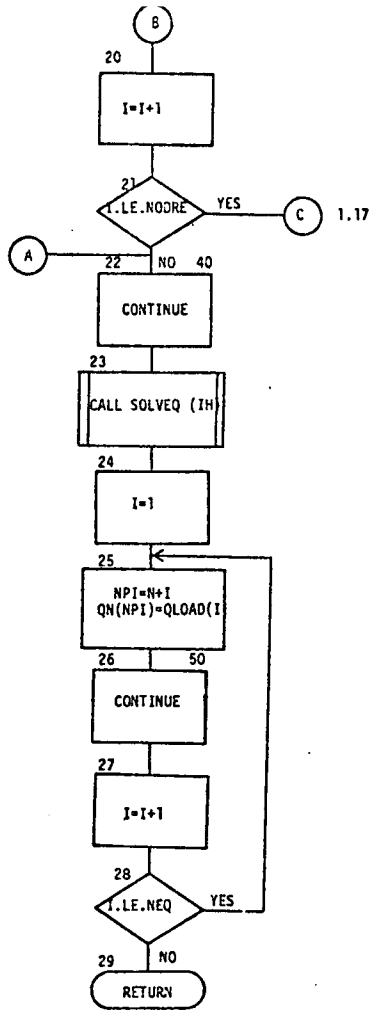




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HOU8QN  
Detail  
1 of 1



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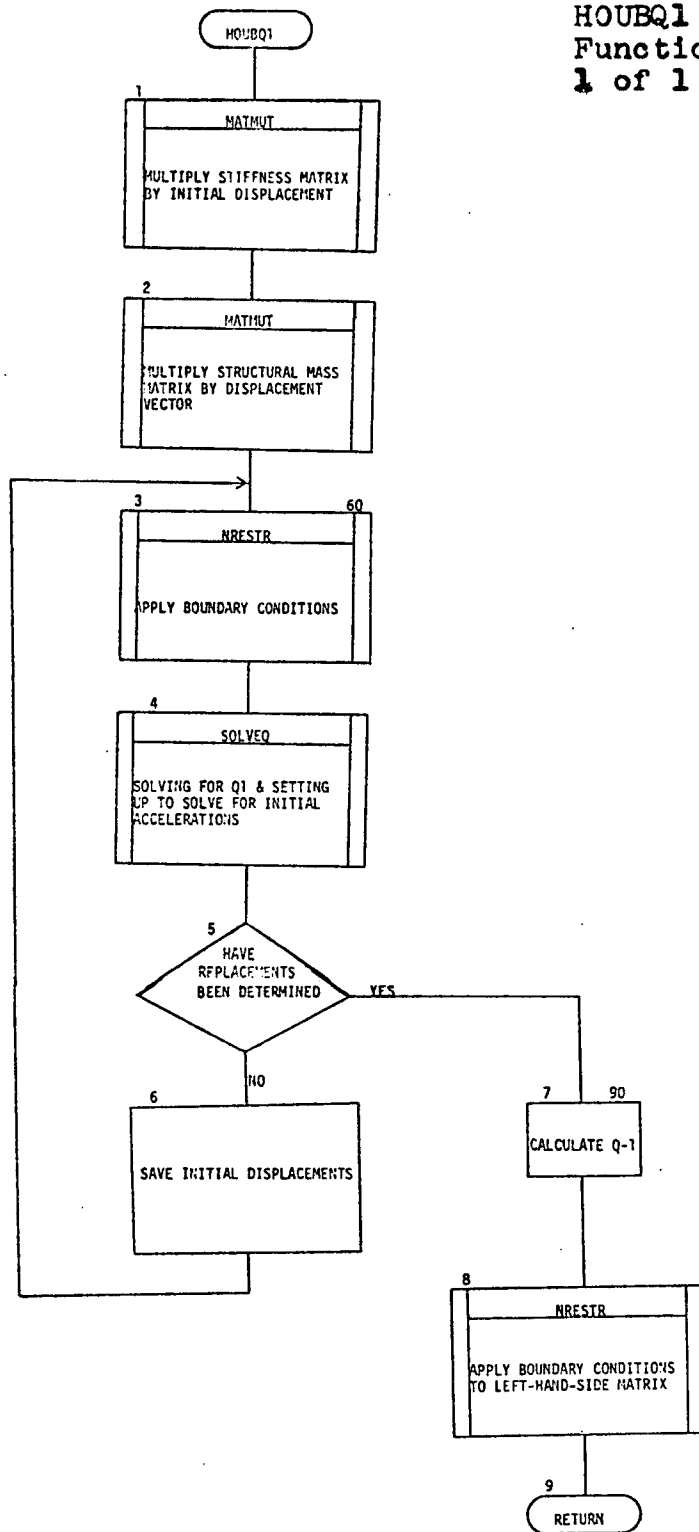
C(HOUBQ1) DYN08962
C DYN08964
C DESCRIPTION - TO CALCULATE THE DISPLACEMENTS AT THE END DYN08966
C OF THE FIRST TIME STEP ONLY AND SET UP THE COEFFICIENT DYN08968
C MATRICES FOR USE IN SUBSEQUENT STEPS. DYN08970
C DYN08972
C INPUT ARGUMENTS. DYN08974
C DELTE = TIME INCREMENT USED IN SOLVING THE EQUATIONS OF MOTION DYN08976
C OF THE SHELL. DYN08978
C DT2 = THE SQUARE OF THE TIME INCREMENT. DYN08980
C FORCE = MATRIX OF GENERALIZED FORCES DUE TO EXTERNAL LOADS AND DYN08982
C TEMPERATURES. DYN08984
C IH = HARMONIC KEY. DYN08986
C IRSTRT = INPUT CONSTANT WHICH INDICATES IF THE PROGRAM IS BEING DYN08988
C RESTARTED. DYN08990
C KY = RESTART KEY. DYN08992
C QN = DISPLACEMENTS AT TIME INCREMENT (N-1) UP TO STATEMENT 20 DYN08994
C AFTER STATEMENT 30 THIS MATRIX HAS BEEN CHANGED TO DYN08996
C THE DISPLACEMENTS AT TIME STEP (N). DYN08998
C TIME = CURRENT TIME. DYN09000
C TO = INITIAL TIME. DYN09002
C T1 = STOP TIME. DYN09004
C XN = STRUCTURAL STIFFNESS MATRIX AS READ FROM INPUT TAPE. DYN09006
C AFTER THE FIRST TIME STEP, THIS MATRIX IS REPLACED BY A DYN09008
C COMBINATION OF THE MASS AND STIFFNESS MATRICES. DYN09010
C DYN09012
C EXTERNALS. DYN09014
C CALLED BY DYN09016
C SETUP DYN09018
C CALLS DYN09020
C MATMUT DYN09022
C NRESTR DYN09024
C SOLVEQ DYN09026
C DYN09028
0001 SUBROUTINE HOUBQ1 (KY,IH) DYN09030
0002 IMPLICIT REAL*8 (A-H,O-Z) DYN09032
0003 COMMON /SLVEEQ/ XN(6550),QLOAD(204) DYN09034
0004 COMMON /QS/ QN(1020),QN1(1020),FORCE(2040),QP(1020),QP1(1020), DYN09036
1 QN2(1020) DYN09038
0005 COMMON /RSTRNT/ NCDRES,NCLOSE,LK(204) DYN09040
0006 COMMON /CONST/ NH,NELEMS,NNODES,NSIZE,NPRNTQ,NEQ,NEQT,N,NN,NHNS, DYN09042
1 DT2,NPRNTL,NPRNTF,IDELF,IDCOE DYN09044
0007 COMMON /PS/ XP(6550) DYN09046
0008 COMMON /TMFT/ TOTIME,DELTE,TIME,TO,T1 DYN09050
0009 COMMON /RESTR/ IRSTRT,NPRNT,NPRNTI,ITP,TIMEP,DELTEP DYN09060
0010 DIMENSION QLOAD1(1020) DYN09070
C DYN09080
0011 IFLAG=0 DYN09130

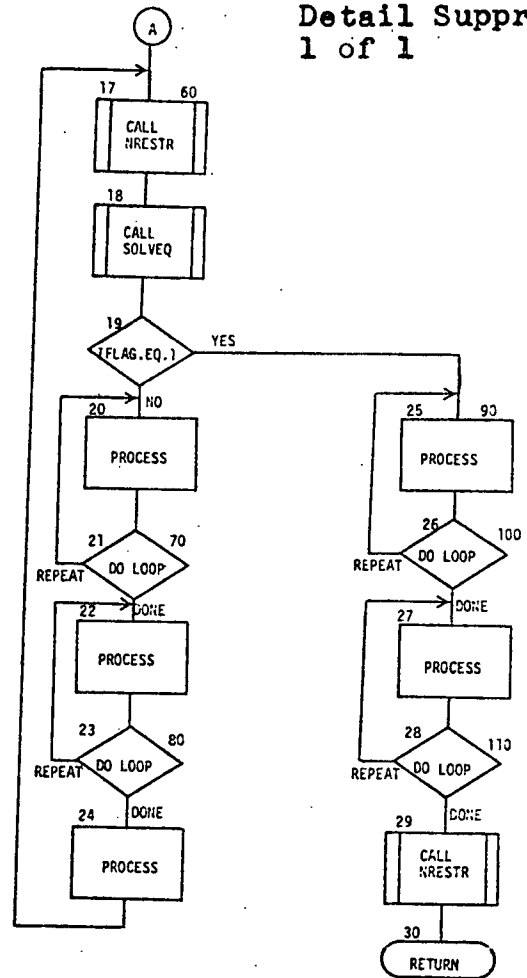
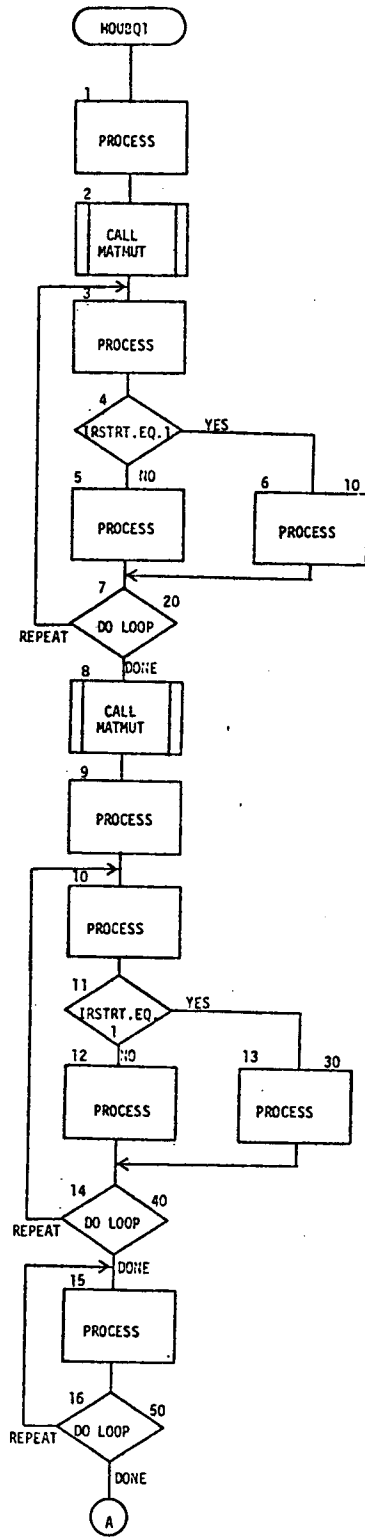
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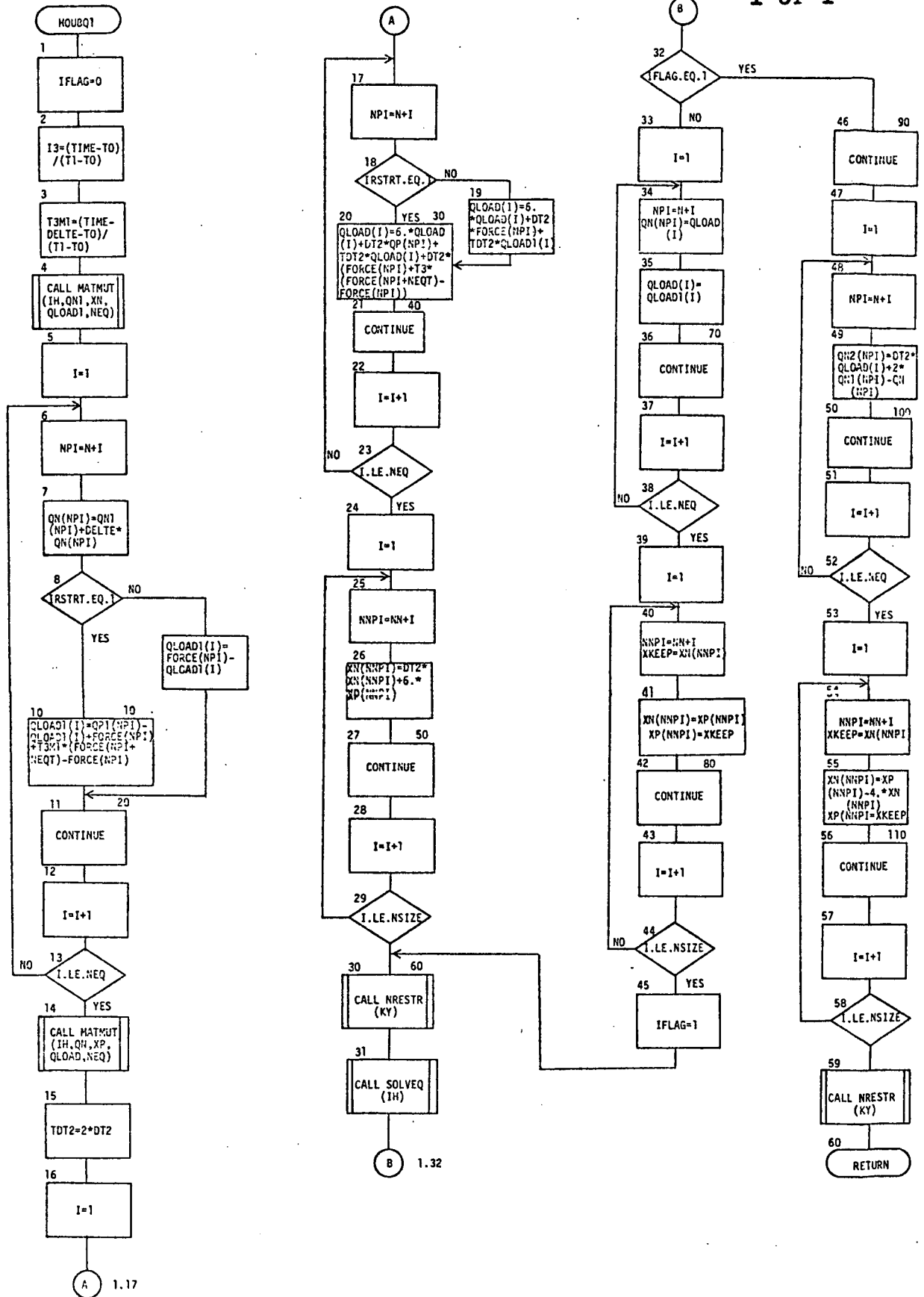




	C		DYN09518
0044		DO 80 I=1,NSIZE	DYNC9520
0045		NNPI=NN+I	DYNC9530
0046		XKEEP=XN(NNPI)	DYN09540
0047		XN(NNPI)=XP(NNPI)	DYN09550
0048		XP(NNPI)=XKEEP	DYNC9560
0049		80 CONTINUE	DYNC9570
	C		DYN09573
0050		IFLAG=1	DYN09590
0051		GO TO 60	DYNC9600
0052		90 CONTINUE	DYNC9610
	C1	CALCULATE Q-1	DYN09612
	C		DYN09628
0053		DO 100 I=1,NEQ	DYNC9630
0054		NPI=N+I	DYN09640
0055		QN2(NPI)=DT2*QLOAD(I)+2*QN1(NPI)-QN(NPI)	DYN09650
0056		100 CONTINUE	DYN09660
	C		DYN09663
	C		DYN09668
0057		DO 110 I=1,NSIZE	DYNC9670
0058		NNPI=NN+I	DYN09680
0059		XKEEP=XN(NNPI)	DYN09690
0060		XN(NNPI)=XP(NNPI)-4.*XN(NNPI)	DYN09700
0061		XP(NNPI)=XKEEP	DYN09710
0062		110 CONTINUE	DYNC9720
	C		DYNC9723
	C1	APPLY BOUNDARY CONDITIONS TO LEFT-HAND-SIDE MATRIX	DYN09725
0063		CALL NRESTR (KY)	DYNC9740
0064		RETURN	DYNC9750
0065		END	DYN09760







CE(INPUT)		DYNO0912
C		DYNO0914
C	DESCRIPTION - TO PERFORM THE MAJOR INPUT FUNCTIONS FOR THE	DYNO0916
C	DYNASOR PROGRAM. IT READS ALL CASE CONTROL PARAMETERS	DYNO0918
C	AND ALL DATA CARD TYPES. SOME DATA REFINEMENT	DYNO0920
C	FUNCTIONS, SUCH AS CALCULATION OF THE TRIGONOMETRIC	DYNO0922
C	INTEGRALS REQUIRED TO CALCULATE THE GENERALIZED	DYNO0924
C	NONLINEAR LOADS, ARE PERFORMED. THE FOLLOWING	DYNO0926
C	QUANTITIES CAN BE READ IN - FOURIER HARMONICS,	DYNO0928
C	MASS AND STIFFNESS MATRICES, NODAL RESTRAINTS,	DYNO0930
C	INITIAL CONDITIONS, SHELL STRUCTURAL DATA, THERMAL	DYNO0932
C	LOADS, THERMAL EXPANSION COEFFICIENTS, CONCENTRATED	DYNO0934
C	RING LOADS, TEMPERATURE DISTRIBUTIONS AND GRADIENTS.	DYNO0936
C		DYNO0938
C	INPUT ARGUMENTS.	DYNO0940
C	KEY = FLAG GOVERNING =1) DATA INPUT FROM TAPE AND CARDS	DYNO0942
C	=2) BYPASSING DATA INPUT DURING RESTART	DYNO0944
C	=3) PERIODIC OUTPUT OF RESTART DATA TO	DYNO0946
C	TAPE.	DYNO0948
C		DYNO0950
C	OUTPUT ARGUMENTS.	DYNO0952
C	ALS = MATRIX OF COEFFICIENTS OF THERMAL EXPANSION IN THE	DYNO0954
C	MERIDIANAL DIRECTION FOR THE ELEMENTS.	DYNO0956
C	ALT = MATRIX OF COEFFICIENTS OF THERMAL EXPANSION IN THE	DYNO0958
C	CIRCUMFERENTIAL DIRECTION FOR THE ELEMENTS.	DYNO0960
C	COSM = MATRIX WHOSE ELEMENTS ARE THE COSINE OF PHI AT THE	DYNO0962
C	MIDDLE OF EACH ELEMENT.	DYNO0964
C	DTH = MATRIX OF FOURIER COEFFICIENTS FOR THE CIRCUMFERENTIAL	DYNO0966
C	TEMPERATURE GRADIENT DISTRIBUTION.	DYNO0968
C	FORCE = MATRIX OF GENERALIZED FORCES DUE TO EXTERNAL LOADS AND	DYNO0970
C	TEMPERATURES.	DYNO0972
C	LK = MATRIX INDICATING THE NODAL RESTRAINTS WHICH ARE APPLIED	DYNO0974
C	ON THE SHELL.	DYNO0976
C	NEQ = NUMBER OF EQUILIBRIUM EQUATIONS PER HARMONIC.	DYNO0978
C	NEQT = TOTAL NUMBER OF EQUILIBRIUM EQUATIONS FOR ALL HARMONICS.	DYNO0980
C	NHNS = LENGTH OF STRUCTURAL STIFFNESS OR MASS MATRIX FOR ALL	DYNO0982
C	HARMONICS STORED IN VECTOR FORM.	DYNO0984
C	NNODES = TOTAL NUMBER OF NODES, EQUAL TO (NFLEMS + 1).	DYNO0986
C	NSIZE = THE NUMBER OF TERMS IN THE STRUCTURAL STIFFNESS OR MASS	DYNO0988
C	MATRIX (IN VECTOR FORM) FOR A PARTICULAR HARMONIC.	DYNO0990
C	QN = INITIAL NODAL VELOCITIES.	DYNO0992
C	QN1 = DISPLACEMENTS AT TIME INCREMENT (N-2) BEFORE STATEMENT 10	DYNO0994
C	AND AT TIME INCREMENT (N-1) AFTER STATEMENT 20.	DYNO0996
C	QP = Q - PARTIAL DERIVATIVE OF U-NL WITH RESPECT TO LOWER	DYNO0998
C	CASE Q AT TIME STEP (N-1).	DYNO1000
C	QP1 = Q - PARTIAL DERIVATIVE OF U-NL WITH RESPECT TO LOWER CASE	DYNO1002
C	Q AT TIME STEP (N-2).	DYNO1004
C	SINM = SINE OF PHI AT THE MIDDLE OF THE ELEMENTS.	DYNO1006

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C      TH      = MATRIX WHOSE ELEMENTS ARE THE FOURIER COEFFICIENTS      DYN01008
C              OF THE CIRCUMFERENTIAL TEMPERATURE DISTRIBUTION.        DYN01010
C      TO      = INITIAL TIME.                                           DYN01012
C      T1      = STOP TIME.                                              DYN01014
C
C      EXTERNALS.
C      CALLED BY
C              MAIN
C              SETUP
C      CALLS
C              TRI4OR
C              NLTERM
C              FRCS
C              THCOE
C              TFORCE
C
0001      SUBROUTINE INPUT (KEY)
0002      IMPLICIT REAL*8 (A-H,O-Z)
0003      COMMON /CHALS/ AL(167),CHECK(8,8)
0004      COMMON /SLVEEQ/ XN(6550),QLOAD(204)
0005      COMMON /QS/ QN(1020),QN1(1020),FORCE(2040),QP(1020),QP1(1020),
1          QN2(1020)
0006      COMMON /RSTRNT/ NODRES,NCLOSE,LK(204)
0007      COMMON /CONST/ NH,NELEMS,NNODES,NSIZE,NPRNTQ,NEQ,NEQT,N,NN,NHNS,
1          DT2,NPRNTL,NPRNTF,IDEFL,IDCOE
0008      COMMON /PS/ XP(6550)
0009      COMMON /TMFT/ TOTIME,DELTE,TIME,TO,T1
0010      COMMON /GEOM/ FNU1(50),FNU2(50),E1(50),E2(50),G(50),T(50),
1          SINE(51),COSINE(51),SINM(50),COSM(50),R(50),PH(50),
1          PHP(50),ARCL(50)
0011      COMMON /THETAS/ THETA(20),NTHETA,NCLCST,NSTRSS
0012      COMMON /PRINT/ IPRINT,NOIT,LL
0013      COMMON /HARM/ NHP,IHARM(5)
0014      COMMON /RESTR/ IRSTRT,NPRNT,NPRNTI,ITP,TIMEP,DELTEP
0015      COMMON /THER/ TH(50,5,2),DTH(50,5,2),ALS(50),ALT(50)
0016      COMMON /THCON/ ITSELF,ITCOE,NPRNTH
0017      COMMON /CYCLE/ ITAM
0018      COMMON /TAPES/ NT,ND,NS
0019      COMMON /RZ/ RO(51),Z(51)
0020      DIMENSION COMENT(20), JUNK(20), TH1(5), DTH1(5)
0021      DIMENSION DUM(1310)
0022      EQUIVALENCE (DUM(1),XN(1)), (XN(1),COMENT(1))
0023      DOUBLE PRECISION CONSTN,CONSTF,CONST1
0024      DATA CONSTN/8HCONSTANT/,CONST1/8H /
C1      KEY//1(10),2(430),3(710)
0025      GO TO (10,430,710), KEY
C1      READ CASE I.D., CASE CONTROL, AND CIRCUMFERENTIAL ANGLE CARDS
0026      10 CONTINUE
    
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DYN01190

0027		CONSTF=CONST1	DYN01200
0028		WRITE (6,730)	DYN01210
	C2	READ INPUT DATA FOR CARD TYPE II - A	DYN01212
0029		READ (ND,740) NCARDS,NT	DYN01230
	C2	TYPE II-B CARDS LEFT//NO(30)	DYN01232
0030		IF (NCARDS.EQ.0) GO TO 30	DYN01240
	C		DYN01248
0031		DO 20 I=1,NCARDS	DYN01250
	C2	READ INPUT DATA FOR CARD TYPE II - B	DYN01252
0032		READ (ND,750) (COMENT(J),J=1,20)	DYN01270
0033		WRITE (6,800) (COMENT(J),J=1,20)	DYN01280
0034		20 CONTINUE	DYN01282
	C		DYN01285
0035		30 CONTINUE	DYN01290
	C2	READ INPUT DATA FOR CARD TYPE III	DYN01292
0036		READ (ND,760) TOTIME,DELTE,IRSTRT,INCRST,NCLOSE,ITELF,NPRNTQ,	DYN01320
	1	IPRINT,NCLCST,NSTRSS,NPRNT,NPRNIT,NPRNTL,NPRNTF,NPRNTH,	DYN01322
	1	NPRNMS	DYN01330
	C2	READ INPUT DATA FOR CARD TYPE IV	DYN01332
0037		READ (ND,770) NTHETA,(THETA(I),I=1,NTHETA)	DYN01340
0038		NOIT=1	DYN01350
	C1	READ FOURIER HARMONICS CARD IF NO RESTART	DYN01352
0039		IF (IRSTRT.EQ.0) READ (ND,780) NH,(IHARM(I),I=1,NH)	DYN01370
0040		KEYRS=0	DYN01380
0041		IF (IRSTRT.EQ.1) KEYRS=1	DYN01390
0042		40 CONTINUE	DYN01400
	C2	READ INFORMATION AND SHELL DESCRIPTION FROM INPUT TAPE	DYN01402
0043		REWIND NT	DYN01440
0044		READ (NT) NCARDS,JUNK	DYN01450
0045		IF (NCARDS.EQ.0) GO TO 60	DYN01460
0046		WRITE (6,790)	DYN01470
	C		DYN01478
0047		DO 50 K=1,NCARDS	DYN01480
0048		READ (NT) (COMENT(J),J=1,20)	DYN01490
0049		IF (KEYRS.EQ.1) GO TO 50	DYN01500
0050		WRITE (6,800) (COMENT(J),J=1,20)	DYN01510
0051		50 CONTINUE	DYN01520
	C		DYN01523
0052		60 READ (NT) NHP,NELEMS,JUNK	DYN01530
0053		IF (KEYRS.EQ.0) GO TO 110	DYN01540
	C2	READ ADDITIONAL INFORMATION FROM INPUT TAPE FOR RESTART OPERATION	DYN01542
	C		DYN01578
0054		DO 70 K=1,NELEMS	DYN01580
0055		READ (NT) (DUM(I),I=1,230)	DYN01590
0056		70 CONTINUE	DYN01592
	C		DYN01595
0057		J=6*NELEMS	DYN01600
0058		READ (NT) (DUM(I),I=1,J)	DYN01610



0059	C	DO 90 K=1,NELEMS	DYN01618
0060		IF (K.EQ.NELEMS) GO TO 80	DYN01620
0061		READ (NT) (DUM(I),I=1,6)	DYN01630
0062		GO TO 90	DYN01640
0063	80	READ (NT) (DUM(I),I=1,8)	DYN01650
0064	90	CONTINUE	DYN01660
	C		DYN01670
0065		J=2*(NELEMS+1)	DYN01673
0066		READ (NT) (DUM(I),I=1,J)	DYN01680
0067		J=2*NHP	DYN01690
0068		NSIZE=10+26*NELEMS	DYN01700
	C		DYN01710
0069		DO 100 K=1,J	DYN01718
0070		READ (NT) (DUM(I),I=1,NSIZE)	DYN01720
0071	100	CONTINUE	DYN01730
	C		DYN01732
0072		READ (NT) NH, (IHARM(I),I=1,NH), JUNK	DYN01735
0073		KEYRS=0	DYN01740
0074		GO TO 40	DYN01750
0075	110	CONTINUE	DYN01760
0076		WRITE (6,810) TOTIME,DELTE,IRSTRT,INCRST,NPRNT,NPRNIT,NPRNTQ,	DYN01770
	1	IPRINT,NCLCST,NSTRSS,NPRNTL,NPRNTF,NPRNTH,NT,NS,ND,	DYN01780
	2	NCLOSE,ITELF,NELFMS,NPRNMS,NH, (IHARM(I),I=1,NH)	DYN01790
0077		WRITE (6,820) NTHETA, (THETA(I),I=1,NTHETA)	DYN01800
	C		DYN01810
	C2	READ IN QN1 INITIAL DISPLACEMENTS Q0	DYN01820
	C2C	QN INITIAL VELOCITIES Q0DOT	DYN01822
	C2C	XN STIFFNESS MATRIX K	DYN01824
	C2C	KP MASS MATRIX M	DYN01826
	C2C	FORCE LOADS	DYN01828
	C2C	NODRES NUMBER OF NODAL RESTRAINTS	DYN01830
	C2C	LK LOCATION OF RESTRAINTS	DYN01832
	C		DYN01834
0078		PI=3.14159	DYN01900
0079		RAD=PI/180.	DYN01910
	C		DYN01920
0080		DO 120 I=1,NTHETA	DYN01928
0081		THETA(I)=THETA(I)*RAD	DYN01930
0082	120	CONTINUE	DYN01940
	C		DYN01950
0083		NNODES=NELEMS+1	DYN01953
0084		NEQ=4*NNODES	DYN01960
0085		NEQT=NH*NEQ	DYN01970
0086		NSIZE=10+26*NELEMS	DYN01980
0087		NHNS=NH*NSIZE	DYN01990
0088		DT2=DELTE**2	DYN02000
	C		DYN02010
			DYN02018

0089	DO 130 I=1,NEQT	DYN02020
0090	QN(I)=0.0	DYN02030
0091	QN1(I)=0.0	DYN02040
0092	130 CONTINUE	DYN02050
	C	DYN02053
	C1 RESTART//YES(250)	DYN02055
0093	IF (IRSTRT.EQ.1) GO TO 250	DYN02060
	C1 READ NODAL RESTRAINT AND INITIAL CONDITIONS CARDS.	DYN02062
	C READ INPUT DATA FOR CARD TYPE VI - A	DYN02100
0094	READ (ND,780) NODRES	DYN02110
0095	WRITE (6,830) NODRES	DYN02120
0096	IF (NODRES.EQ.0) GO TO 150	DYN02130
	C	DYN02138
0097	DO 140 I=1,NODRES	DYN02140
	C READ INPUT DATA FOR CARD TYPE VI - B	DYNG2150
0098	READ (ND,780) NP,NDIRCT	DYN02160
0099	WRITE (6,840) NP,NDIRCT	DYN02170
0100	LK(I)=4*(NP-1)+NDIRCT	DYN02180
0101	140 CONTINUE	DYN02182
	C	DYN02185
0102	150 LK(NODRES+1)=3	DYN02190
0103	LK(NODRES+2)=4	DYN02200
	C2 READ AND PRINT INITIAL VELOCITIES AND DISPLACEMENTS	DYN02202
	C READ INPUT DATA FOR CARD TYPE VII - A	DYN02240
0104	READ (ND,780) IQN,IQN1	DYN02250
0105	IF (IQN.EQ.0) GO TO 190	DYN02260
	C	DYN02268
0106	DO 180 IH=1,NH	DYN02270
0107	N=NEQ*(IH-1)	DYN02280
	C2 READ INPUT DATA FOR CARD TYPE VII - B	DYN02282
0108	160 READ (ND,850) IN1,IN2,Q1,Q2,Q3,Q4	DYN02300
	C	DYN02308
0109	DO 170 INODE=IN1,IN2	DYN02310
0110	IFLAG=4*(INODE-1)+N	DYN02320
0111	QN(IFLAG+1)=Q1	DYN02330
0112	QN(IFLAG+2)=Q2	DYN02340
0113	QN(IFLAG+3)=Q3	DYN02350
0114	QN(IFLAG+4)=Q4	DYN02360
0115	170 CONTINUE	DYN02362
	C	DYN02365
0116	IF (IN2.NE.NNODES) GO TO 160	DYN02370
0117	180 CONTINUE	DYN02380
	C	DYN02383
0118	190 CONTINUE	DYN02390
0119	IF (IQN1.EQ.0) GO TO 230	DYN02400
	C	DYN02408
0120	DO 220 IH=1,NH	DYN02410
0121	N=NEQ*(IH-1)	DYN02420

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0122      C2      READ INPUT DATA FOR CARD TYPE VII - C          DYN02422
          200    READ (ND,850) IN1,IN2,Q1,Q2,Q3,Q4                DYN02440
          C      DYN02448
0123      C      DO 210 INODE=IN1,IN2                            DYN02450
0124      C      IFLAG=4*(INODE-1)+N                            DYN02460
0125      C      QN1(IFLAG+1)=Q1                                 DYN02470
0126      C      QN1(IFLAG+2)=Q2                                 DYN02480
0127      C      QN1(IFLAG+3)=Q3                                 DYN02490
0128      C      QN1(IFLAG+4)=Q4                                 DYN02500
0129      C      210    CONTINUE                                  DYN02502
          C      DYN02505
0130      C      IF (IN2.NE.NNODES) GO TO 200                    DYN02510
0131      C      220 CONTINUE                                     DYN02520
          C      DYN02523
0132      C      230 CONTINUE                                     DYN02530
0133      C      WRITE (6,860)                                     DYN02540
          C      DYN02548
0134      C      DO 240 I=1,NH                                    DYN02550
0135      C      IQ=NEQ*(I-1)                                    DYN02560
          C      DYN02568
0136      C      DO 240 II=1,NNODES                               DYN02570
0137      C      IX=4*(II-1)                                     DYN02580
0138      C      WRITE (6,870) II,IHARM(I),QN(IQ+IX+1),QN(IQ+IX+2), DYN02590
          1      QN(IQ+IX+3),QN(IQ+IX+4),QN1(IQ+IX+1),QN1(IQ+IX+2), DYN02600
          1      QN1(IQ+IX+3),QN1(IQ+IX+4)                        DYN02601
0139      C      240 CONTINUE                                     DYN02610
          C      DYN02613
0140      C      250 CONTINUE                                     DYN02620
0141      C      REWIND NS                                         DYN02630
          C2     READ STRUCTURAL DATA FOR SHELL                  DYN02632
          C      DYN02638
0142      C      DO 260 II=1,NELEMS                               DYN02640
0143      C      READ (NT) ((CHECK(I,J),I=1,8),J=1,8),(AL(I),I=1,166) DYN02650
0144      C      WRITE (NS) ((CHECK(I,J),I=1,8),J=1,8),(AL(I),I=1,166) DYN02660
0145      C      260 CONTINUE                                     DYN02670
          C      DYN02673
0146      C      READ (NT) (FNU1(I),I=1,NELEMS),(FNU2(I),I=1,NELEMS),(E1(I),I=1, DYN02680
          1      NELEMS),(E2(I),I=1,NELEMS),(G(I),I=1,NELEMS),(T(I),I=1, DYN02690
          1      NELEMS)                                         DYN02692
          C      DYN02698
0147      C      DO 280 I=1,NELEMS                                DYN02700
0148      C      IF (I.EQ.NELEMS) GO TO 270                      DYN02710
0149      C      READ (NT) R(I),PH(I),PHP(I),ARCL(I),SINE(I),COSINE(I) DYN02720
0150      C      GO TO 280                                        DYN02730
0151      C      270    READ (NT) R(I),PH(I),PHP(I),ARCL(I),SINE(I),COSINE(I), DYN02740
          1      SINE(I+1),COSINE(I+1)                            DYN02750
0152      C      280 CONTINUE                                     DYN02760
          C      DYN02763

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0153		READ (NT) (RO(I),I=1,NNODES),(Z(I),I=1,NNODES)	DYN02770
	C		DYN02778
0154		DO 290 I=1,NELEMS	DYN02780
0155		COSM(I)=DCOS(PH(I))	DYN02790
0156		SINM(I)=DSIN(PH(I))	DYN02800
0157		290 CONTINUE	DYN02810
	C1	INCLUDE THERMAL LOADS//NO(320)	DYN02815
	C		DYN02817
0158		IF (ITELF.NE.1) GO TO 320	DYN02820
	C1	READ COEFFICIENTS OF THERMAL EXPANSION CARDS	DYN02822
	C2	READ INPUT DATA FOR CARD. TYPE VIII	DYN02824
0159		300 READ (ND,880) IELM1,IELM2,ALSI1,ALTI1	DYN02870
	C		DYN02878
0160		DO 310 IELM=IELM1,IELM2	DYN02880
0161		ALS(IELM)=ALSI1	DYN02890
0162		ALT(IELM)=ALTI1	DYN02900
0163		310 CONTINUE	DYN02902
	C		DYN02905
0164		IF (IELM2.NE.NELEMS) GO TO 300	DYN02910
0165		GO TO 340	DYN02920
	C		DYN02928
0166		320 DO 330 IELM=1,NELEMS	DYN02930
0167		ALS(IELM)=0.0	DYN02940
0168		ALT(IELM)=0.0	DYN02950
0169		330 CONTINUE	DYN02960
	C		DYN02963
0170		340 CONTINUE	DYN02970
	C2	PRINT ELEMENT PROPERTIES AND DESIRED STIFFNESS AND MASS MATRICES	DYN02972
0171		WRITE (6,890)	DYN03010
0172		WRITE (6,900) (I,ALS(I),ALT(I),E1(I),E2(I),FNU1(I),FNU2(I),G(I),	DYN03020
	1	R(I),T(I),ARCL(I),PH(I),PHP(I),I=1,NELEMS)	DYN03030
	C		DYN03038
0173		DO 370 IH=1,NHP	DYN03040
	C		DYN03048
0174		DO 350 JH=1,NH	DYN03050
0175		IF (IH-1.EQ.IHARM(JH)) GO TO 360	DYN03060
0176		350 CONTINUE	DYN03070
	C		DYN03073
0177		READ (NT) (FORCE(I),I=1,NSIZE)	DYN03080
0178		READ (NT) (FORCE(I),I=1,NSIZE)	DYN03090
0179		GO TO 370	DYN03100
0180		360 CONTINUE	DYN03110
0181		NN=NSIZE*(JH-1)	DYN03120
0182		READ (NT) (XN(I+NN),I=1,NSIZE)	DYN03130
0183		READ (NT) (XP(I+NN),I=1,NSIZE)	DYN03140
0184		IF (NPRNMS.EQ.0) GO TO 370	DYN03150
0185		WRITE (6,910) IHARM(JH)	DYN03160
0186		WRITE (6,920) (XN(I+NN),I=1,NSIZE)	DYN03170



0217	C	DO 420 I=1,NELEMS	DYN03608
			DYN03610
	C		DYN03618
0218		DO 420 J=1,NH	DYN03620
0219		TH(I,J,1)=0.0	DYN03630
0220		TH(I,J,2)=0.0	DYN03640
0221		DTH(I,J,1)=0.0	DYN03650
0222		DTH(I,J,2)=0.0	DYN03660
0223		420 CONTINUE	DYN03670
	C		DYN03673
	C1	FIRST TIME THROUGH AND CONSTANT FORCES//NO(440)	DYN03675
0224		430 IF (CONSTF.EQ.CONSTN.AND.KEY.NE.1) GO TO 440	DYN03710
0225		GO TO 470	DYN03720
	C1	UPDATE FORCE AND THERMAL MATRICES	DYN03722
	C		DYN03728
0226		440 DO 450 I=1,NEQT	DYN03730
0227		FORCE(I)=FORCE(I+NEQT)	DYN03740
0228		450 CONTINUE	DYN03742
	C		DYN03745
	C		DYN03748
0229		DO 460 I=1,NELEMS	DYN03750
	C		DYN03758
0230		DO 460 J=1,NH	DYN03760
0231		TH(I,J,1)=TH(I,J,2)	DYN03770
0232		DTH(I,J,1)=DTH(I,J,2)	DYN03780
0233		460 CONTINUE	DYN03782
	C		DYNC3785
0234		T0=T1	DYN03790
0235		T1=TOTIME	DYN03800
0236		RETURN	DYN03810
0237		470 CONTINUE	DYN03820
	C		DYN03828
0238		DO 480 I=1,NEQT	DYN03830
0239		FORCE(I)=FORCE(I+NEQT)	DYN03840
0240		FORCE(I+NEQT)=0.0	DYN03850
0241		480 CONTINUE	DYN03860
	C		DYN03863
	C		DYN03868
0242		DO 490 I=1,NELEMS	DYN03870
	C		DYN03878
0243		DO 490 J=1,NH	DYN03880
0244		TH(I,J,1)=TH(I,J,2)	DYN03890
0245		DTH(I,J,1)=DTH(I,J,2)	DYN03900
0246		TH(I,J,2)=0.0	DYN03910
0247		DTH(I,J,2)=0.0	DYN03920
0248		490 CONTINUE	DYN03930
	C		DYN03933
0249		IB=0	DYN03940

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0250          IF (KEY.EQ.2) IB=1                      DYN03950
0251          IF (IB.EQ.1) T0=T1                      DYN03960
          C1    READ LOAD CONTROL CARD                DYN03962
0252          500 READ (ND,950) T1,NCF,IDELF,IDCOE,ITCOE,CONSTF DYN03980
0253          TIM=T1                                  DYN03990
0254          IF (CONSTF.EQ.CONSTN.AND.IB.EQ.0) T1=TOTIME DYN04000
0255          IF (NPRNTL.EQ.0.AND.NPRNTF.EQ.0) GO TO 510 DYN04010
          C2    PRINT LOADING DESCRIPTION              DYN04012
0256          TPRNT=TIM*1000000.                      DYN04050
0257          WRITE (6,960) TPRNT,CONSTF              DYN04060
          C1    CONCENTRATED RING LOADS//NO(550)      DYN04062
0258          510 IF (NCF.EQ.0) GO TO 550              DYN04070
          C1    READ CONCENTRATED RING LOADS          DYN04072
          C                                            DYN04078
0259          DO 540 IH=1,NH                           DYN04080
0260          IH1=IH-1                                 DYN04090
          C    READ INPUT DATA FOR CARD TYPE IX - B - 1 DYN04100
0261          READ (ND,970) NCF1                      DYN04110
0262          IF (NCF1.EQ.0) GO TO 540                 DYN04120
0263          IF (NPRNTL.EQ.0) GO TO 520              DYN04130
0264          WRITE (6,980) IHARM(IH)                 DYN04140
          C    READ INPUT DATA FOR CARD TYPE IX - B - 2 DYN04150
0265          520 READ (ND,970) IN1,IN2,F1,F2,F3,F4    DYN04160
          C                                            DYN04168
0266          DO 530 IN=IN1,IN2                       DYN04170
0267          K=4*IN+NEQ*IH1+IB*NEQT                 DYN04180
0268          FORCE(K-3)=F1                            DYN04190
0269          FORCE(K-2)=F2                            DYN04200
0270          FORCE(K-1)=F3                            DYN04210
0271          FORCE(K)=F4                              DYN04220
0272          IF (NPRNTL.EQ.1) WRITE (6,990) IN,F1,F2,F3,F4 DYN04230
0273          530 CONTINUE                             DYN04240
          C                                            DYN04243
0274          IF (IN2.NE.NNODES) GO TO 520            DYN04250
0275          540 CONTINUE                             DYN04260
          C                                            DYN04263
0276          550 CONTINUE                             DYN04270
          C1    DISTRIBUTED LOADS PRESENT//NO(570)    DYN04272
0277          IF (IDELF.NE.1) GO TO 570              DYN04280
0278          REWIND NS                               DYN04290
          C100  PROCESS ALL ELEMENTS                  DYN04292
          C                                            DYN04298
0279          DO 560 IELM=1,NELEMS                    DYN04300
0280          READ (NS) ((CHECK(I,J),I=1,8),J=1,8),(AL(I),I=1,166) DYN04310
          C1    READ DISTRIBUTED LOADS AND CALCULATE LINEAR GENERALIZED FORCES DYN04312
0281          CALL FRCS (IELM,ALPHK,IB)              DYN04320
0282          560 CONTINUE                             DYN04330
          C                                            DYN04333

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	C1	THERMAL LOADS PRESENT//NO(650)	DYN04335
0283	570	IF (ITELF.EQ.0) GO TO 650	DYN04340
0284		REWIND NS	DYN04380
0285		IBP1=IB+1	DYN04390
	C1	READ THERMAL COEFFICIENTS//NO(600)	DYN04392
0286		IF (ITCOE.EQ.0) GO TO 600	DYN04400
	C1	READ TEMPERATURE DISTRIBUTIONS AND GRADIENTS	DYN04402
0287	580	READ (NO,1000) IELM1,IELM2,(TH1(IH),DTH1(IH),IH=1,NH)	DYN04420
	C		DYN04428
0288		DO 590 IELM=IELM1,IELM2	DYN04430
	C		DYN04438
0289		DO 590 IH=1,NH	DYN04440
0290		DTH(IELM,IH,IBP1)=DTH1(IH)	DYN04450
0291		TH(IELM,IH,IBP1)=TH1(IH)	DYN04460
0292		590 CONTINUE	DYN04462
	C		DYN04465
0293		IF (IELM2.NE.NELEMS) GO TO 580	DYN04470
	C100	PROCESS ALL ELEMENTS	DYN04472
	C		DYN04478
0294	600	DO 620 IELM=1,NELEMS	DYN04480
0295		READ (NS) ((CHECK(I,J),I=1,8),J=1,8),(AL(I),I=1,166)	DYN04490
	C1	CALCULATE THERMAL COEFFICIENTS//NO(610)	DYN04492
0296		IF (ITCOE.EQ.1) GO TO 610	DYN04500
	C1	READS TEMPERATURE AND TEMPERATURE GRADIENTS AND CALCULATES	DYN04502
	C1C	THERMAL FOURIER COEFFICIENTS	DYN04504
0297		CALL THCOE (IELM,IB)	DYN04510
	C1	CALCULATES LINEAR THERMAL LOADS	DYN04512
0298	610	CALL TFORCE (IELM,IB)	DYN04520
0299		620 CONTINUE	DYN04530
	C		DYN04533
0300		IF (NPRNTH.EQ.0) GO TO 640	DYN04540
	C		DYN04548
0301		DO 630 IH=1,NH	DYN04550
0302		WRITE (6,1010) IHARM(IH)	DYN04560
	C		DYN04568
0303		DO 630 IELM=1,NELEMS	DYN04570
0304		WRITE (6,1020) IELM,TH(IELM,IH,IBP1),DTH(IELM,IH,IBP1)	DYN04580
0305		630 CONTINUE	DYN04582
	C		DYN04585
0306		640 CONTINUE	DYN04590
0307	650	IF (NPRNTE.EQ.0) GO TO 670	DYN04600
	C2	PRINT GENERALIZED FORCES FOR EACH HARMONIC	DYN04602
	C		DYN04638
0308		DO 660 IH=1,NH	DYN04640
0309		KK=NEQ*(IH-1)+IB*NEQT	DYN04650
0310		KYP=IHARM(IH)	DYN04660
0311		WRITE (6,1030) KYP	DYN04670
	C		DYN04678



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0312          DO 660 I=1,NNODES                DYN04680
0313          K=KK+4*(I-1)                    DYN04690
0314          WRITE (6,1040) I,FORCE(K+1),FORCE(K+2),FORCE(K+3),FORCE(K+4) DYN04700
0315          660 CONTINUE                      DYN04702
C                                                    DYN04705
0316          670 CONTINUE                      DYN04710
C1          FORCES NOT CONSTANT OR FIRST TIME THROUGH WITH KEY=2//YES(700) DYN04712
0317          IF (CONSTF.NE.CONSTN.OR.IB.EQ.1) GO TO 700 DYN04720
C                                                    DYN04728
0318          DO 680 I=1,NEQT                  DYN04730
0319          FORCE(I+NEQT)=FORCE(I)            DYN04740
0320          680 CONTINUE                      DYN04742
C                                                    DYN04745
C                                                    DYN04748
0321          DO 690 I=1,NELEMS                DYN04750
C                                                    DYN04758
0322          DO 690 J=1,NH                    DYN04760
0323          TH(I,J,2)=TH(I,J,1)              DYN04770
0324          DTH(I,J,2)=DTH(I,J,1)            DYN04780
0325          690 CONTINUE                      DYN04790
C                                                    DYN04793
0326          RETURN                          DYN04800
0327          700 CONTINUE                      DYN04810
0328          IB=IB+1                          DYN04820
C1          KEY = 1 //YES(500)                DYN04822
0329          IF (IB.EQ.1) GO TO 500           DYN04830
0330          RETURN                          DYN04840
C1          WRITE RESTART INFORMATION ON TAPE DYN04842
0331          710 CONTINUE                      DYN04880
0332          NTF=2*NEQT                       DYN04890
0333          WRITE (NT) NH, (IHARM(I), I=1, NH), JUNK DYN04900
0334          WRITE (NT) ITAM, TIME, DELTE, TC, T1, NODRES, (LK(I), I=1, NODRES), DYN04910
0335          1 (FORCE(I), I=1, NTF)            DYN04920
0335          WRITE (NT) (((TH(IELM, IH, IBP1), IELM=1, NELEMS), IH=1, NH), IBP1=1, 2), DYN04930
0335          1 (((DTH(IELM, IH, IBP1), IELM=1, NELEMS), IH=1, NH), IBP1=1, 2) DYN04940
0336          QDC3=1.0/(2.0*DELTE)              DYN04950
0337          QDC2=4.0*QDC3                    DYN04960
0338          QDC1=3.0*QDC3                    DYN04970
C                                                    DYN04978
0339          DO 720 I=1,NEQT                  DYN04980
0340          QP(I)=QDC1*QN(I)-QDC2*QN1(I)+QDC3*QN2(I) DYN04990
0341          720 CONTINUE                      DYN04992
C                                                    DYN04995
0342          WRITE (NT) (QP1(I), I=1, NEQT), (QN(I), I=1, NEQT), (QP(I), I=1, NEQT) DYN05000
0343          TPRNT=TIME*1000000.              DYN05010
0344          WRITE (6,1050) ITAM, TPRNT       DYN05020
0345          RETURN                          DYN05030
C                                                    DYN05040

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0346      730 FORMAT (1H1,38X,44HDYNASOR-II - DYNAMIC NONLINEAR ANALYSIS OF, DYN05050
          1      21H SHELLS OF REVOLUTION//) DYN05060
0347      740 FORMAT (2I5) DYN05070
0348      750 FORMAT (20A4) DYN05080
0349      760 FURMAT (2F10.0,4I5,/,10I5) DYN05090
0350      770 FORMAT (I5,/, (8F10.0)) DYN05100
0351      780 FORMAT (16I5) DYN05110
0352      790 FORMAT (///,2X,46H**SHELL IDENTIFICATION COMMENTS FROM SAMMSOR**) DYN05120
0353      800 FORMAT (/5X,20A4) DYN05130
0354      810 FORMAT (1H1,50X,33HCONTROL CONSTANTS AND COMMENTS// DYN05140
          1      35X,8HTOTIME =,F12.9,22X,7HDELTE =,F13.9/ DYN05142
          1      35X,8HIRSTRT =,I12,22X,8HINCRST =,I12/ DYN05144
          2      35X,7HNPRNT =,I13,22X,8HNPRNIT =,I12/ DYN05146
          2      35X,8HNPRNTQ =,I12,22X,8HIPRINT =,I12/ DYN05148
          3      35X,8HNCLCST =,I12,22X,8HNSTRSS =,I12/ DYN05150
          3      35X,8HNPRNTL =,I12,22X,8HNPRNTF =,I12/ DYN05152
          4      35X,8HNPRNTH =,I12,22X,4HNT =,I16/ DYN05154
          4      35X,4HNS =,I16,22X,4HND =,I16,/ DYN05156
          5      35X,8HNCLOSE =,I12,22X,7HITELF =,I13/ DYN05158
          5      35X,8HNELEMS =,I12,22X,8HNPRNMS =,I12/ DYN05160
          6      35X,4HNNH =,I16/ DYN05162
          6      35X,7HIHARM =,5I11//) DYN05164
0355      820 FORMAT (35X,8HNTHETA =,I12,/35X,7HTHETA =,5F10.2,(/,42X,5F10.2)) DYN05210
0356      830 FORMAT (/////50X,29HNUMBER OF NODAL RESTRAINTS ISI5// DYN05220
          1      52X,9HDIRECTION,12X,7HAPPLIES,// DYN05222
          1      57X,1H1,10X,15HAXIAL RESTRAINT/ DYN05230
          2      57X,1H2,10X,20HTANGENTIAL RESTRAINT,/, DYN05232
          2      57X,1H3,10X,16HRADIAL RESTRAINT/ DYN05240
          3      57X,1H4,10X,17HANGULAR RESTRAINT,// DYN05242
          4      58X,15HNODE DIPECTION//) DYN05250
0357      840 FORMAT (58X,I3,7X,I1) DYN05260
0358      850 FORMAT (2I5,4F10.0) DYN05270
0359      860 FORMAT (1H1,7X,7HINITIAL,29X,10HVELOCITIES,22X,3HAND,19X, DYN05280
          1      13HDISPLACEMENTS// DYN05290
          1      4X,14HNODE HARMONIC,2(3CH AXIAL TANGENTIAL), DYN05300
          2      26H RADIAL ANGULAR )//) DYN05310
0360      870 FORMAT (5X,I2,6X,I2,3X,8D14.4) DYN05320
0361      880 FORMAT (2I5,2F10.0) DYN05330
0362      890 FORMAT (1H1,45X,4HELEMENT ELASTIC AND GEOMETRIC PROPERTIES,/// DYN05340
          1      48H ELEMENT ALPHA--S ALPHA--T E1 E2, DYN05342
          1      11H FNU1 FNU2,7X,1HG,11X,1HR,11X,1HT,9X,4HARCL,9X,2HPPH, DYN05350
          2      10X,3HPPH//) DYN05360
0363      900 FORMAT (3X,I2,2X,4D10.2,2F6.3,6D12.4) DYN05370
0364      910 FORMAT (1H1,38X,15HHARMONIC NUMBER,I5, DYN05380
          1      37H HAS THE FOLLOWING STIFFNESS MATRIX//) DYN05390
0365      920 FORMAT (2X,D16.8,/,2X,2D16.8,/,2X,3D16.8,/,2X,4D16.8,/,2X,5D16.8, DYN05400
          1      2X,6D16.8,/,2X,7D16.8,/,2X,8D16.8, DYN05410
          1      (2X,5D16.8,/2X,6D16.8/2X,7D16.8/2X,8D16.8//) DYN05420

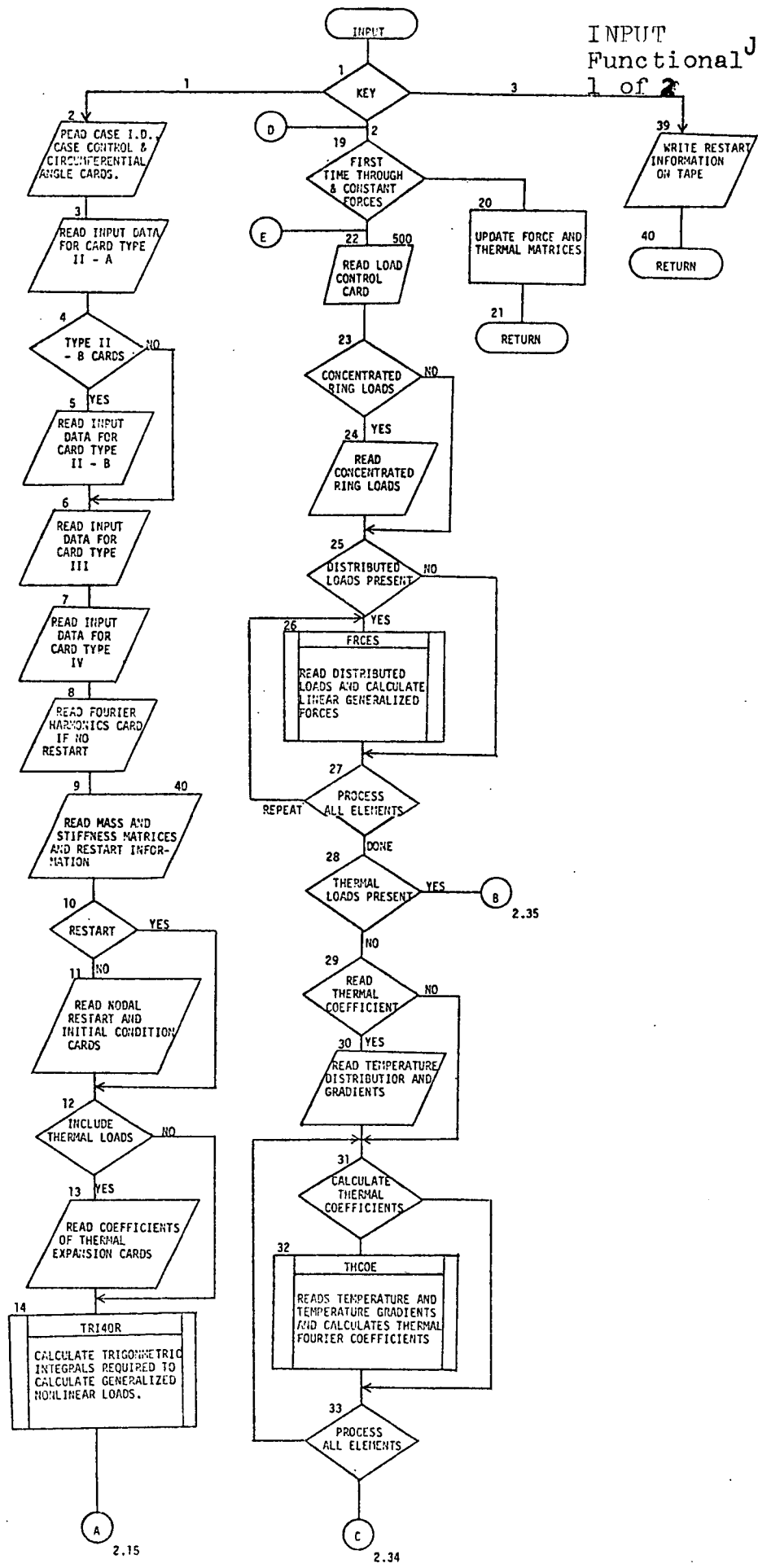
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0366      930 FORMAT (I1,38X,15HHARMONIC NUMBER,I5,          DYN05430
          1          32H HAS THE FOLLOWING MASS MATRIX//)          DYN05440
0367      940 FORMAT (I1,10X,5X,41HTHIS SOLUTION STARTS AFTER TIME INCREMENT, DYN05450
          1          4H NO.,I5,19H WHERE THE TIME WAS,F12.4,13H MICROSECONDS,/ DYN05460
          1          5X,27H AND THE TIME INCREMENT WAS,D12.5////////)          DYN05470
0368      950 FORMAT (F10.0,4I5,A8)          DYN05480
0369      960 FORMAT (40HIFOLLOWING IS LOAD DESCRIPTION AT TIME =,F12.4,          DYN05490
          1          13H MICRCSECONDS,5X,A8)          DYN05500
0370      970 FORMAT (2I5,4F10.0)          DYN05510
0371      980 FORMAT (///20X,30HCONCENTRATED FORCES HARMONIC ,I5//          DYN05520
          1          6X,8HNODE NO.,6X,5HAXIAL,10X,10HTANGENTIAL,10X,6HRADIAL,          DYN05530
          1          7HANGULAR/)          DYN05532
0372      990 FORMAT (I10,4D20.8)          DYN05540
0373      1000 FORMAT (2I5,/, (2F10.0))          DYN05550
0374      1010 FORMAT (I1,25X,39HTEMPERATURE COEFFICIENTS, HARMONIC NO. I3//          DYN05560
          1          10X,11HELEMENT NO.,17X,12HTEMP. COEFF.,12X,          DYN05570
          1          18HTEMP. GRAD. COEFF.///)          DYN05572
0375      1020 FORMAT (I20,2D30.5)          DYN05580
0376      1030 FORMAT (I1,25X,32HGENERALIZED FORCES, HARMONIC NO.,I3,//          DYN05590
          1          6X,8HNODE NO.,6X,5HAXIAL,13X,10HTANGENTIAL,11X,6HRADIAL,          DYN05600
          1          13X,7HANGULAR///)          DYN05602
0377      1040 FORMAT (I9,4D19.8)          DYN05610
0378      1050 FORMAT (I1,10X,5X,42HRESTART INFORMATION FOR TIME INCREMENT NO., DYN05620
          1          I5/          DYN05622
          1          10X,22H CORRESPONDING TO TIME,F12.4,13H MICROSECONDS,/          DYN05630
          2          2X,46H HAS BEEN PLACED ON TAPE FOR USE IN SUBSEQUENT,          DYN05640
          2          5H RUNS//)          DYN05642
0379      END          DYN05650

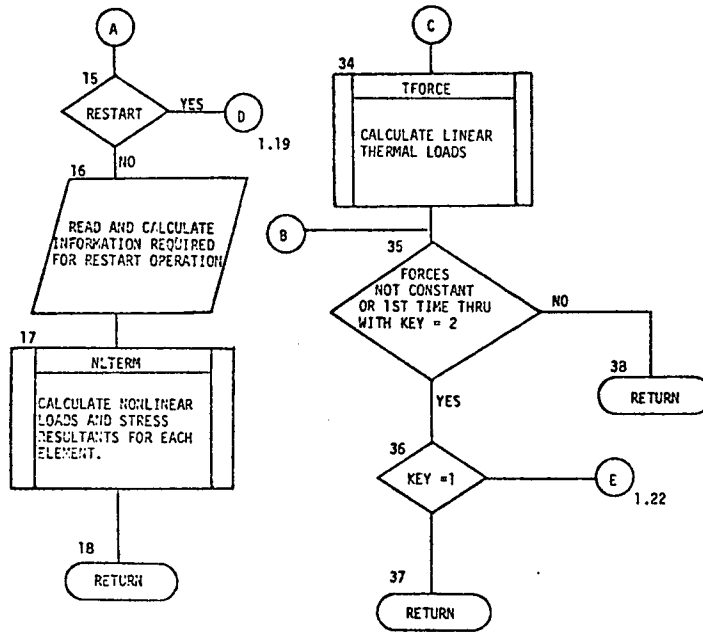
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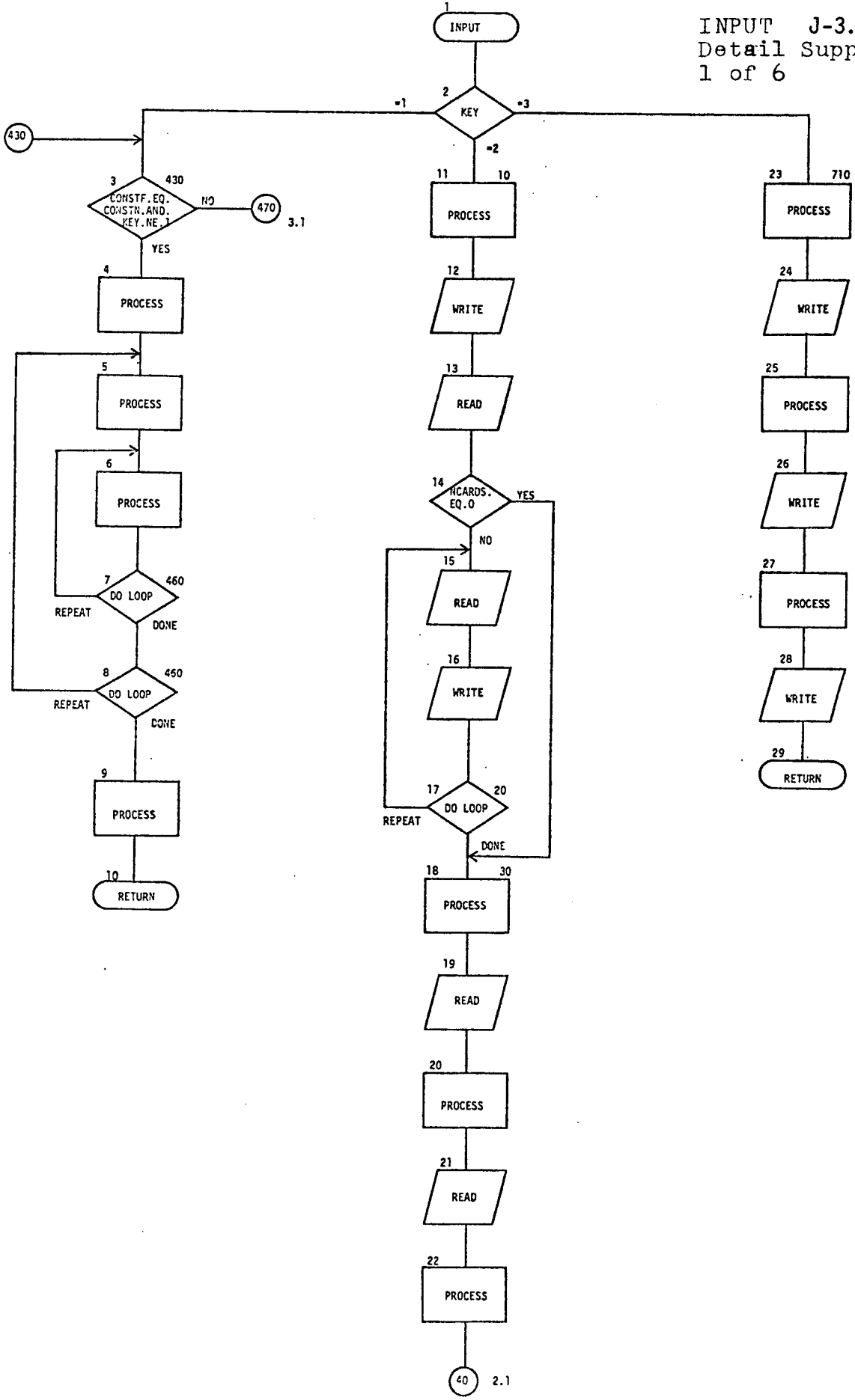


A 2.15

C 2.34

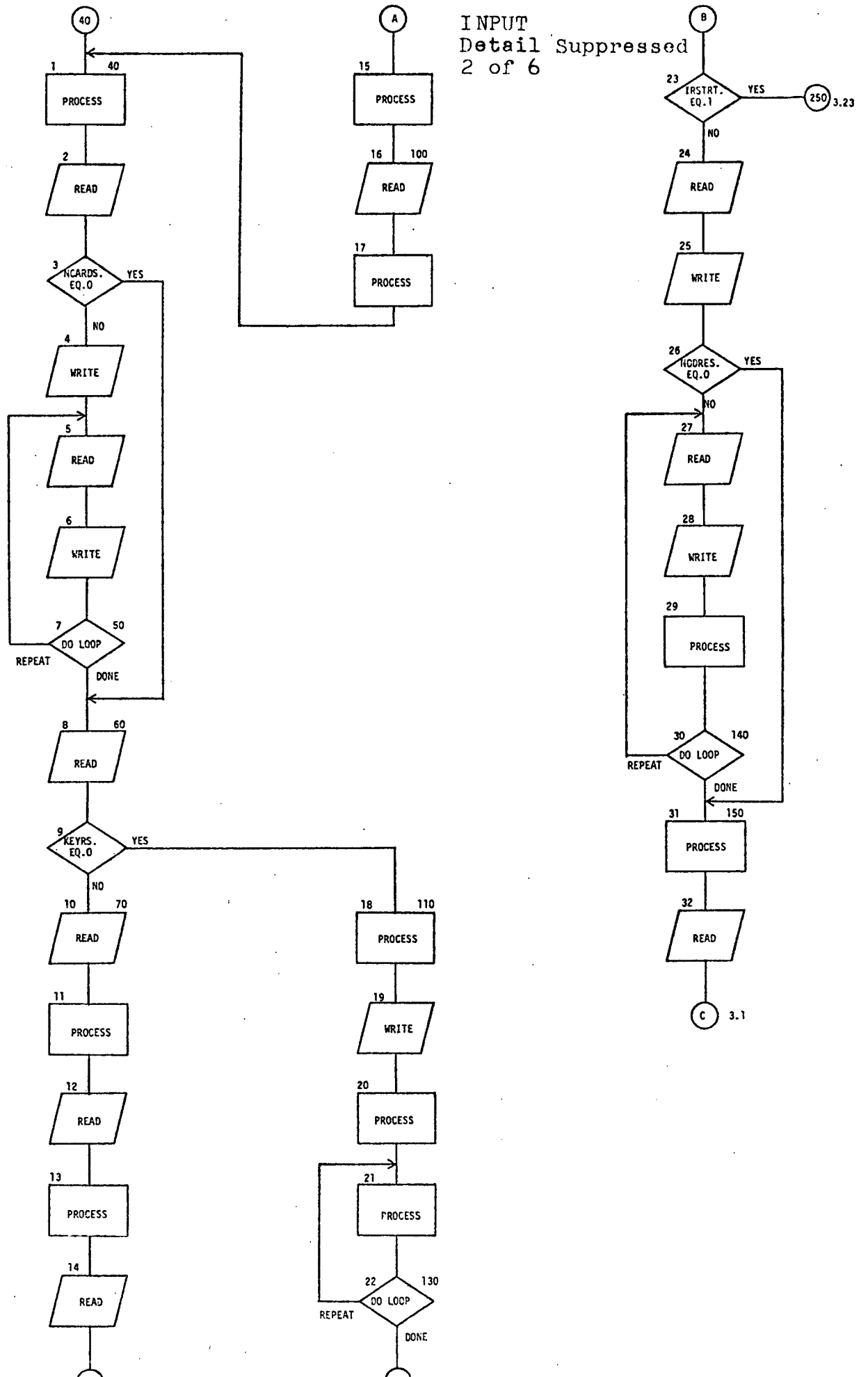
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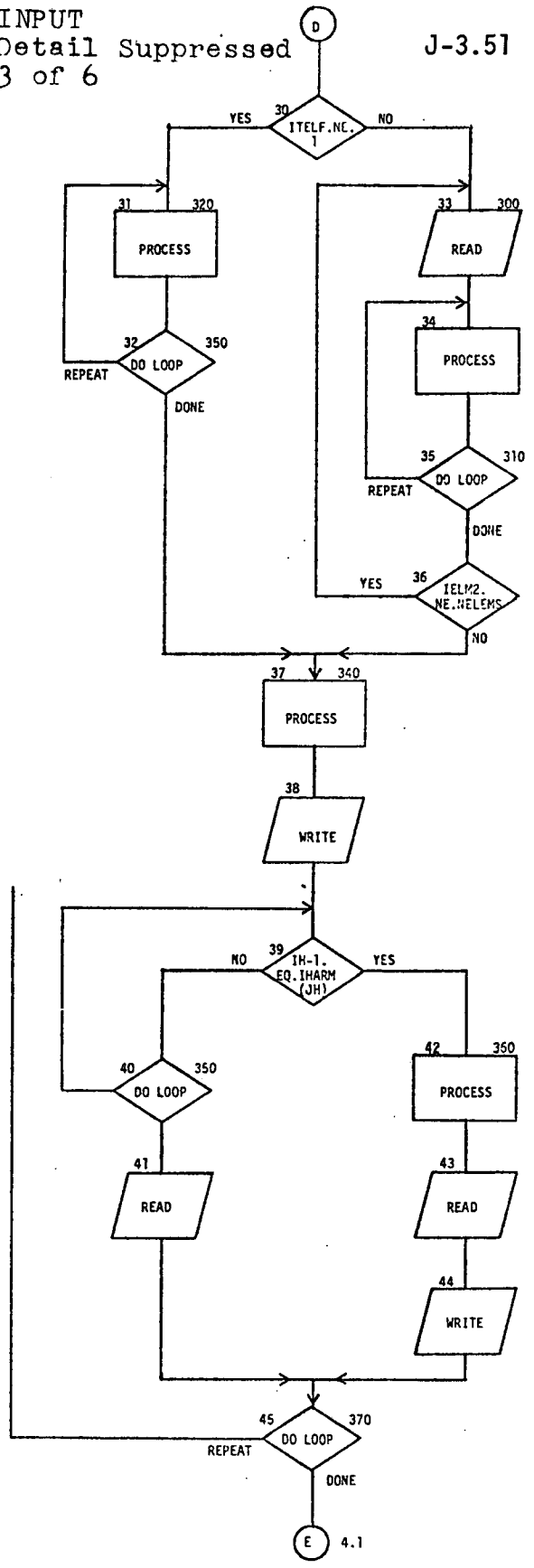
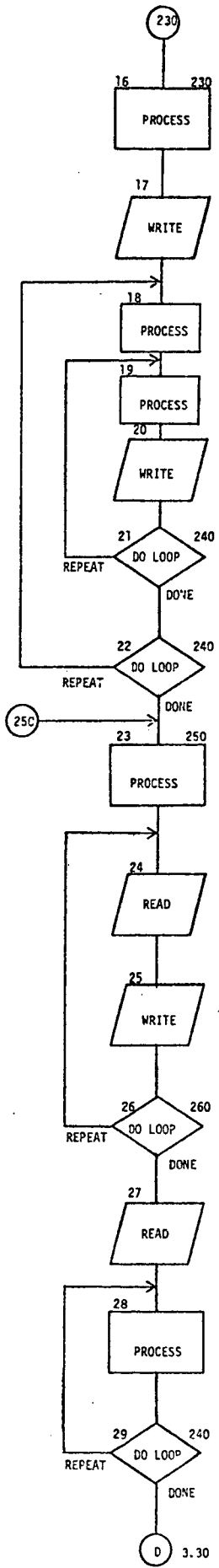
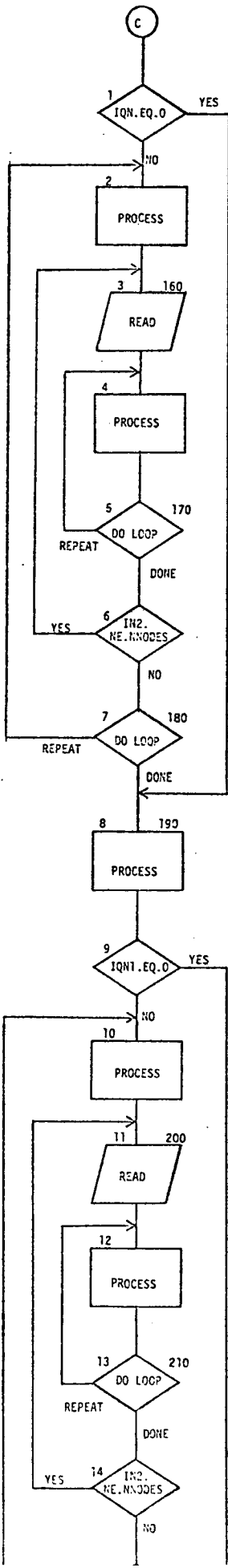
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3 of 6

J-3.51

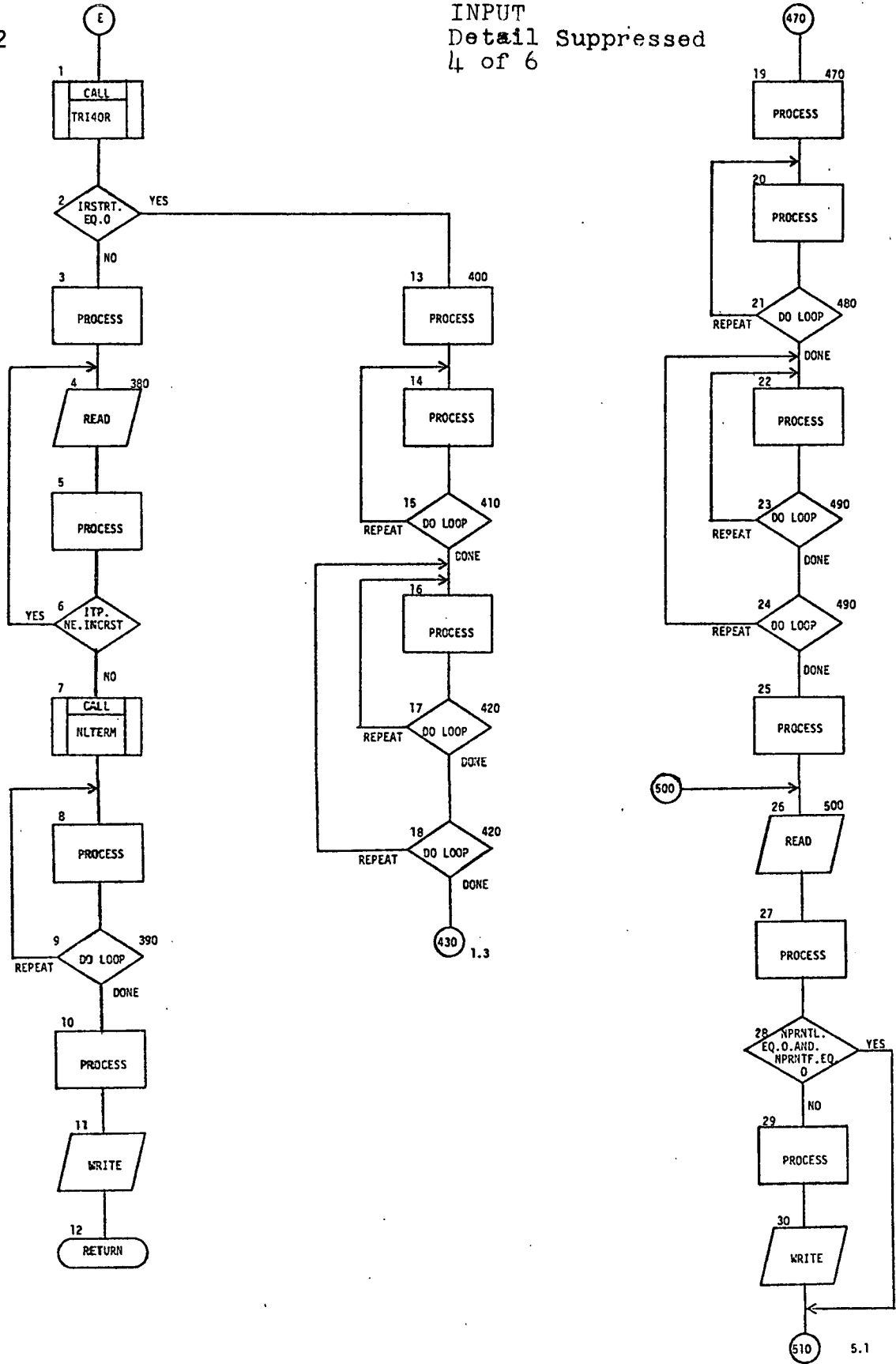


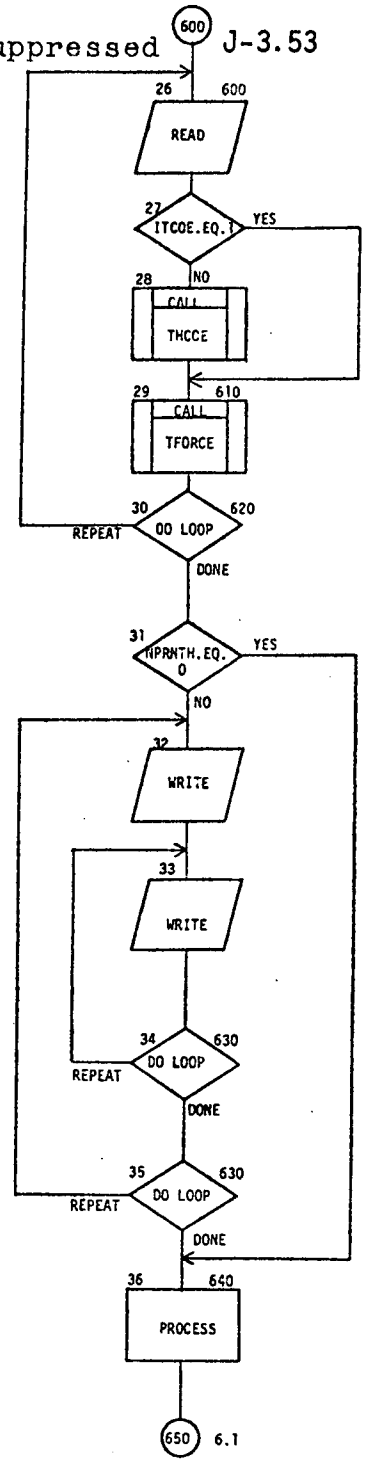
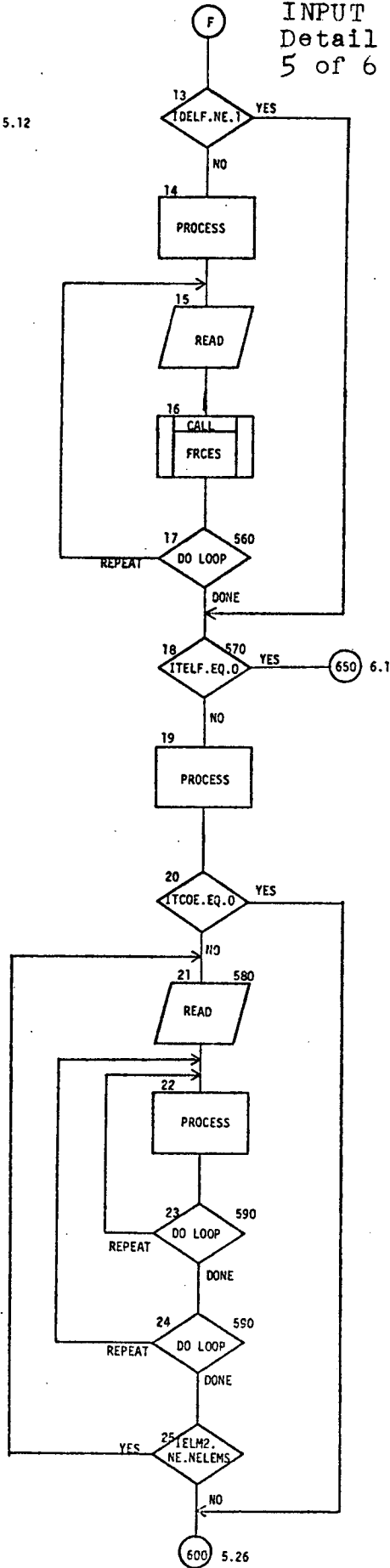
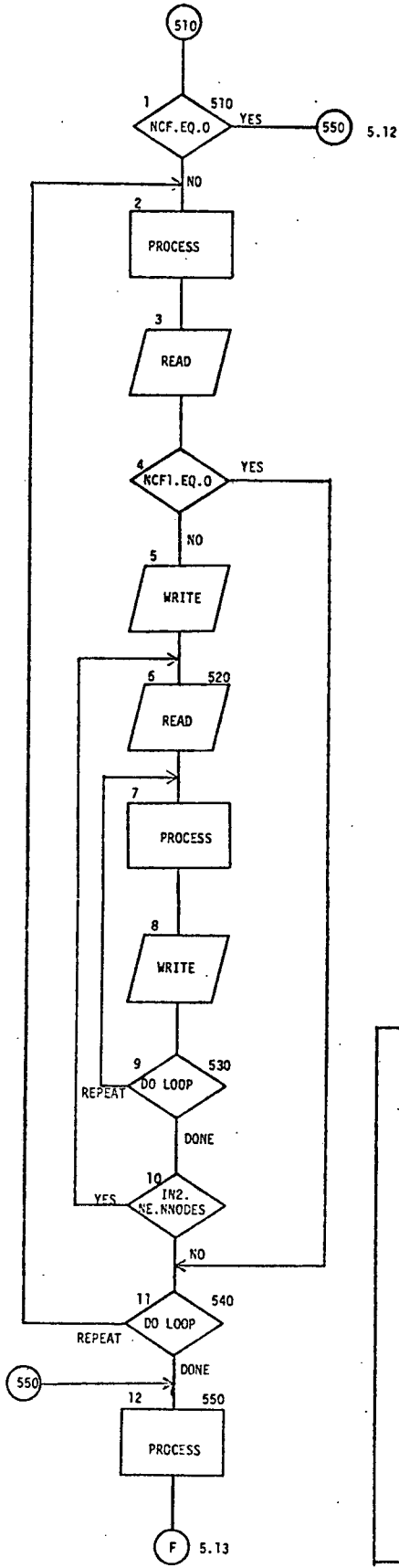
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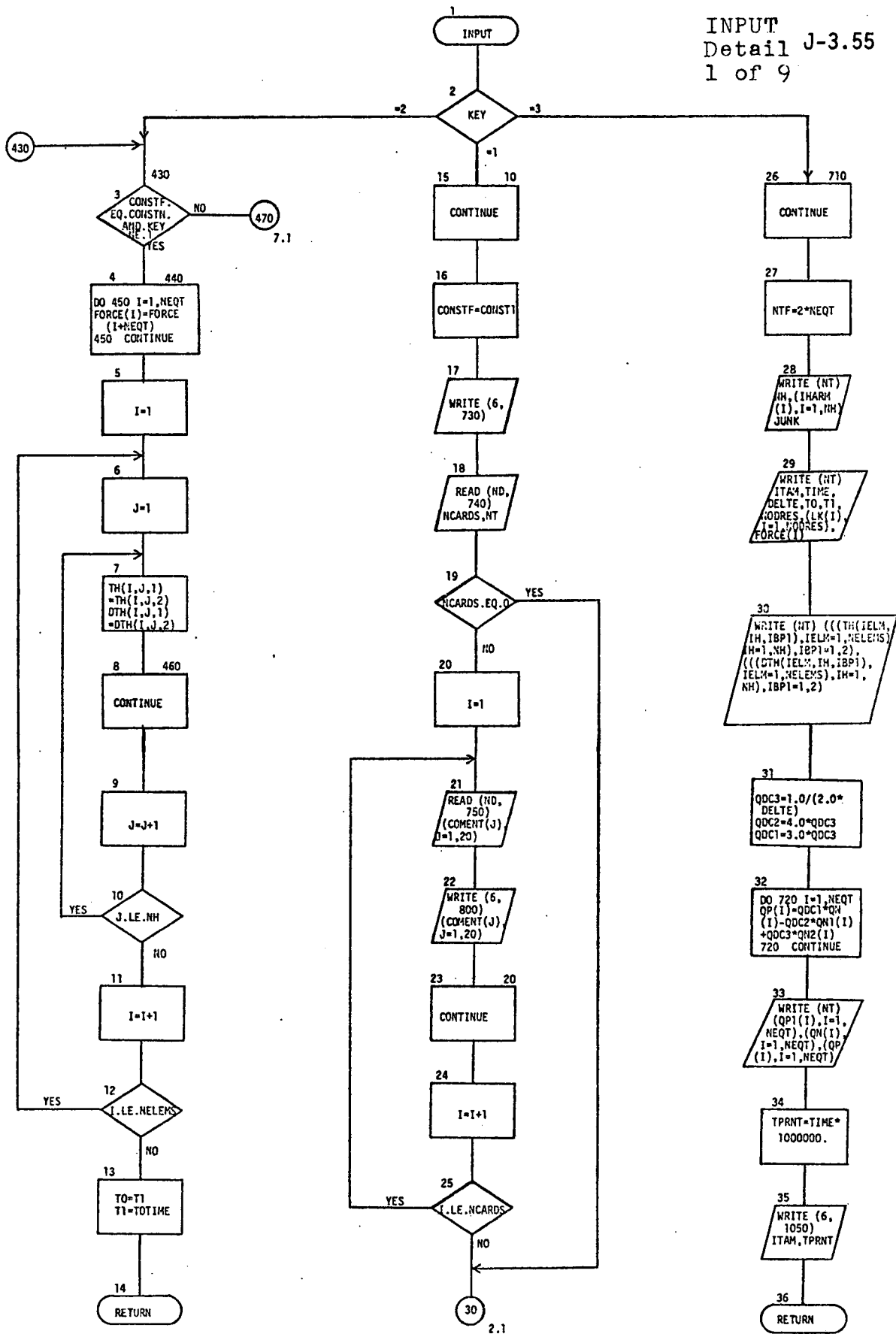
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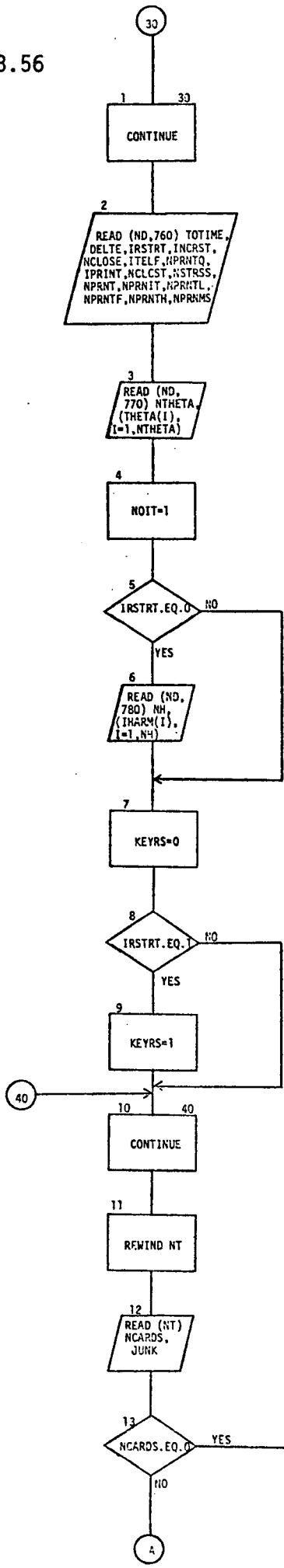




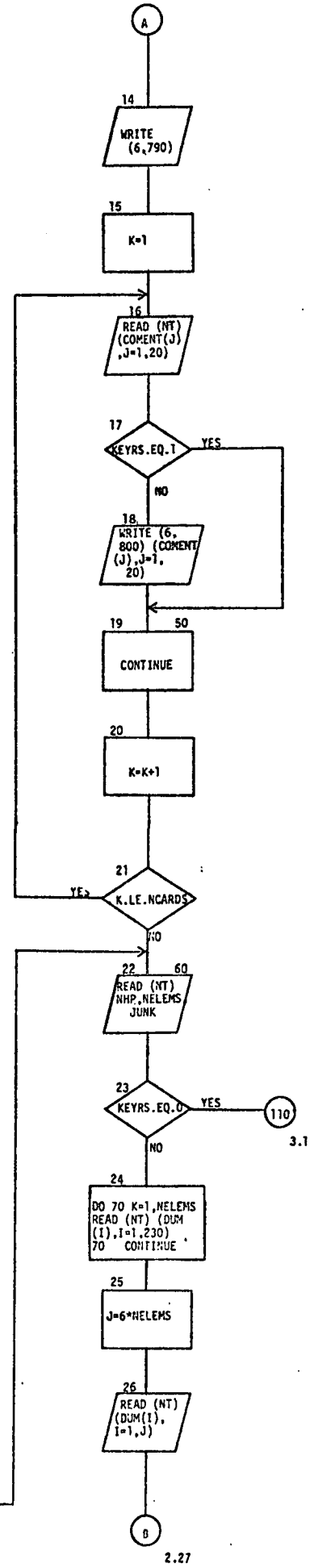




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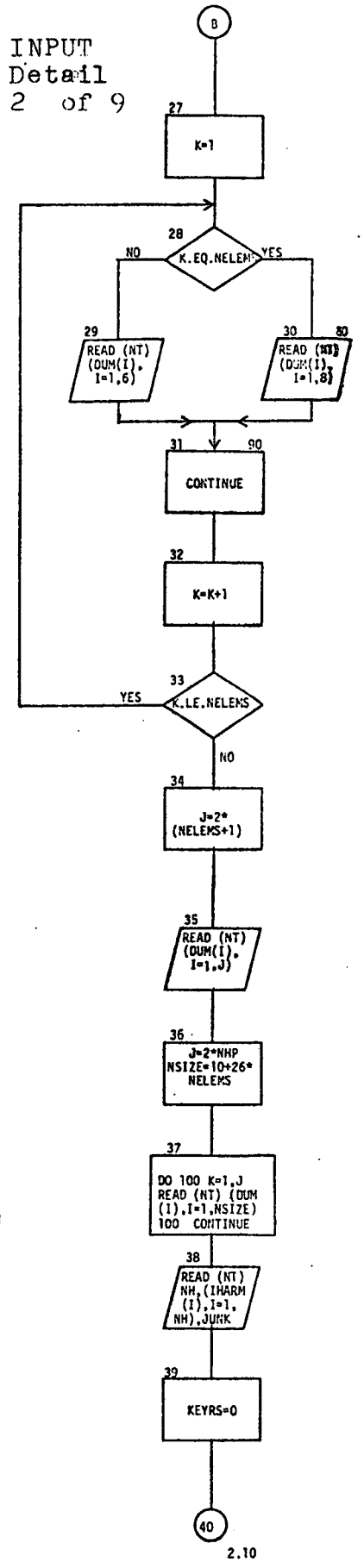


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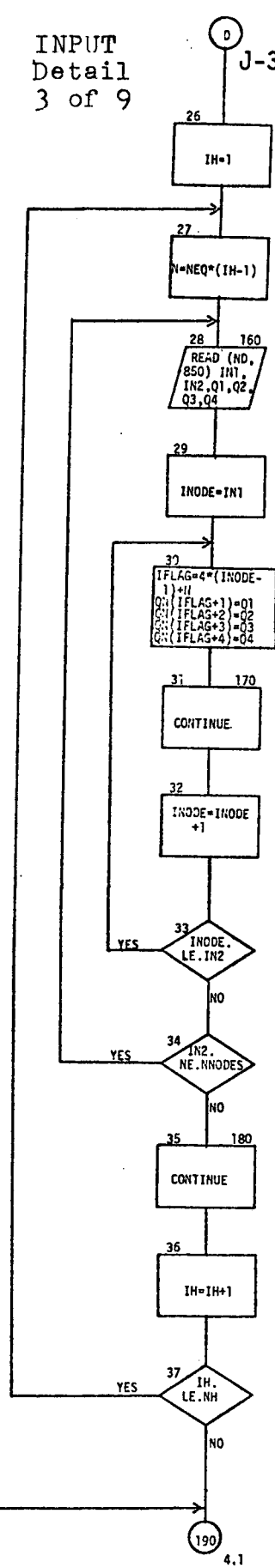
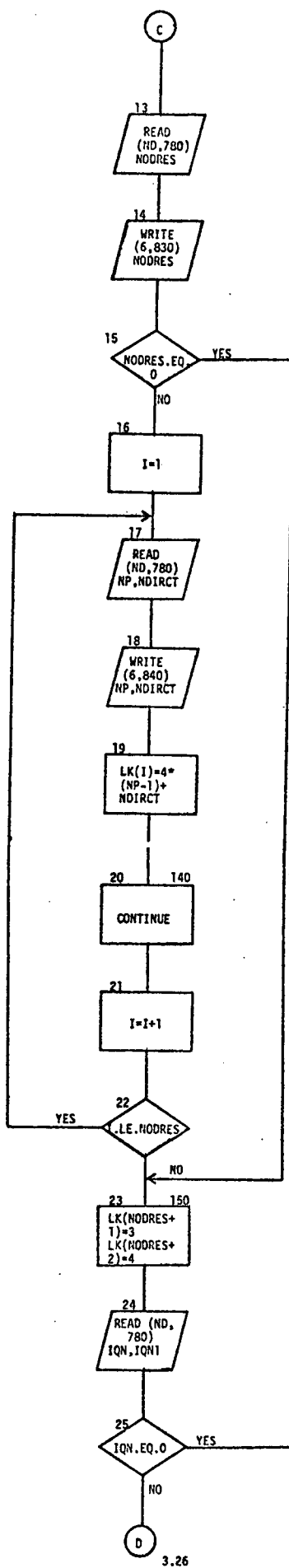
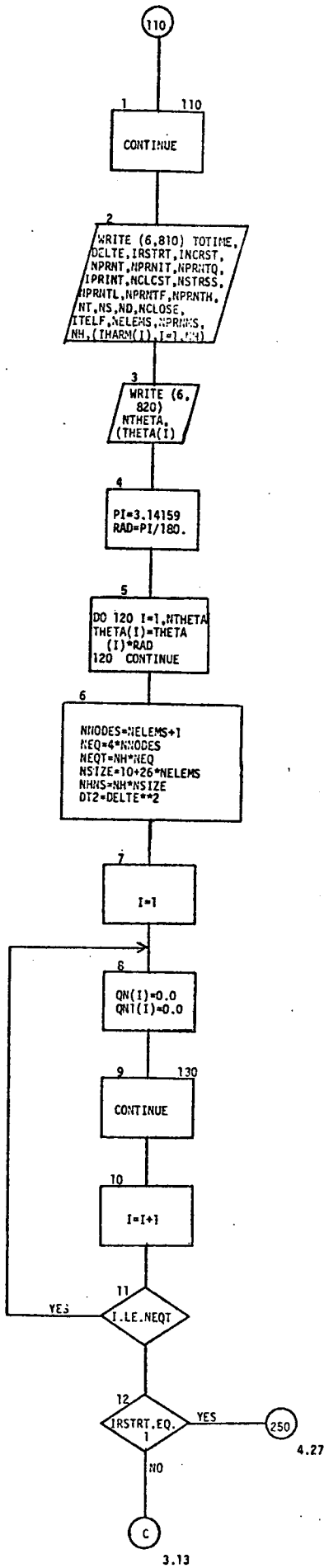


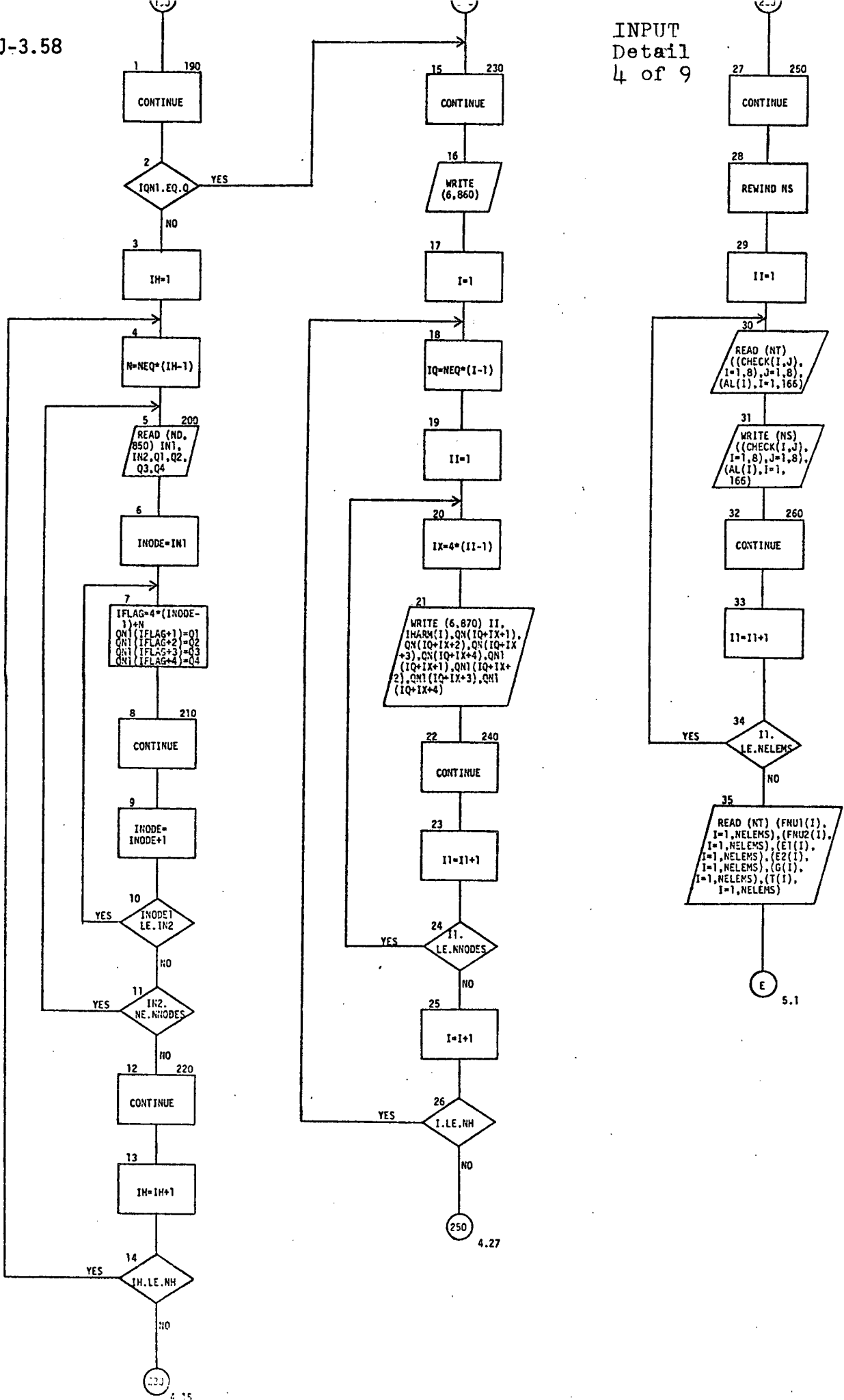
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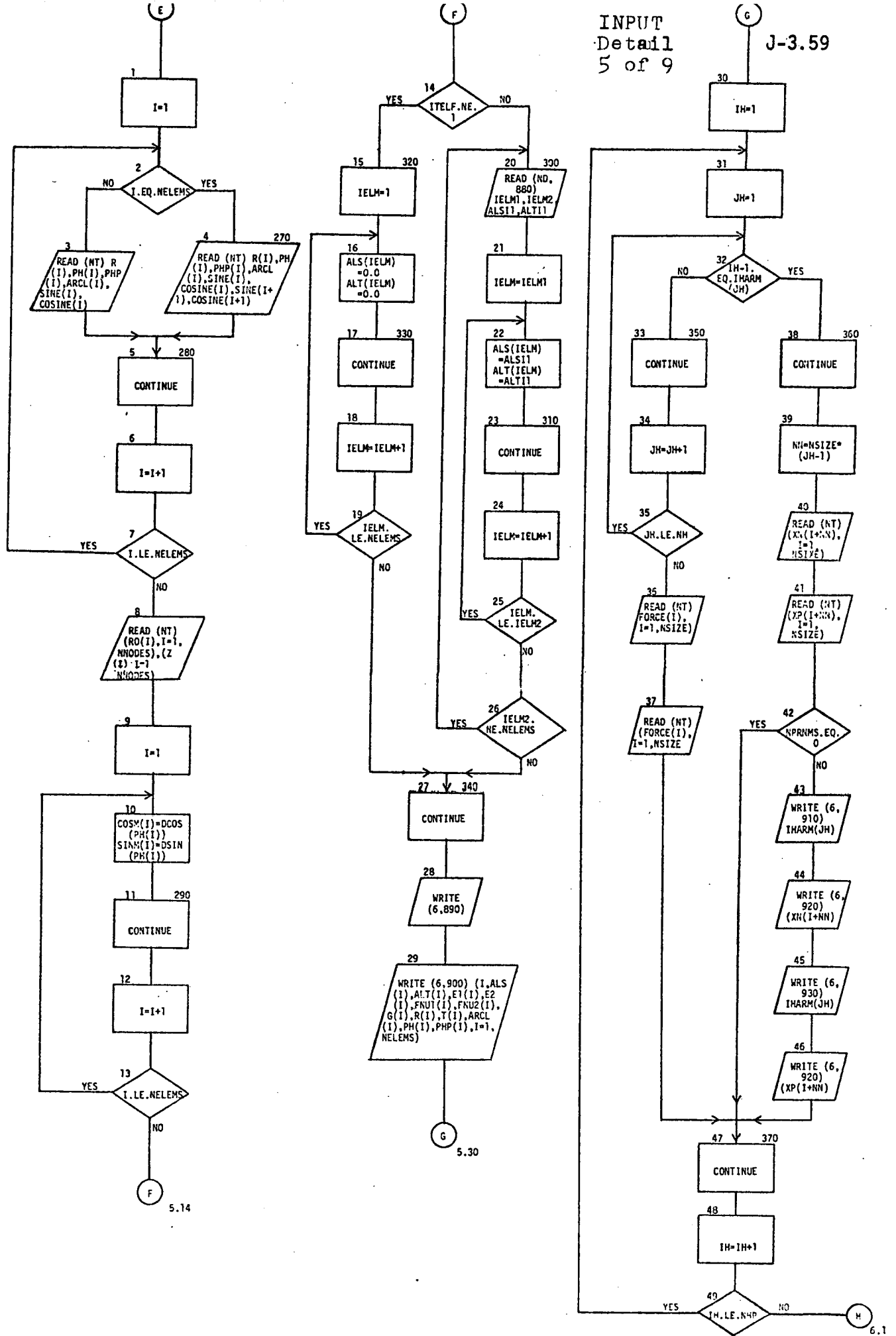
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2.10





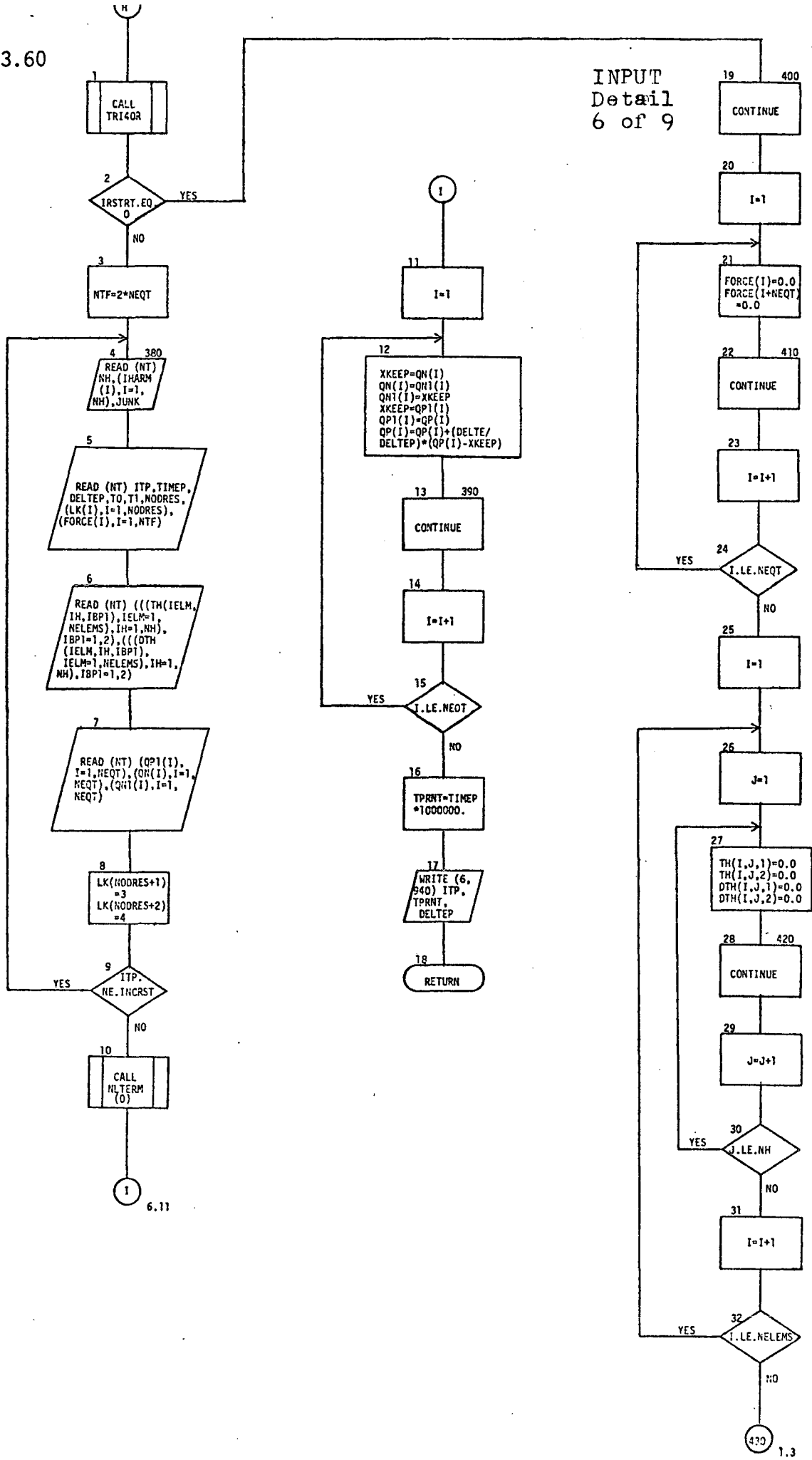


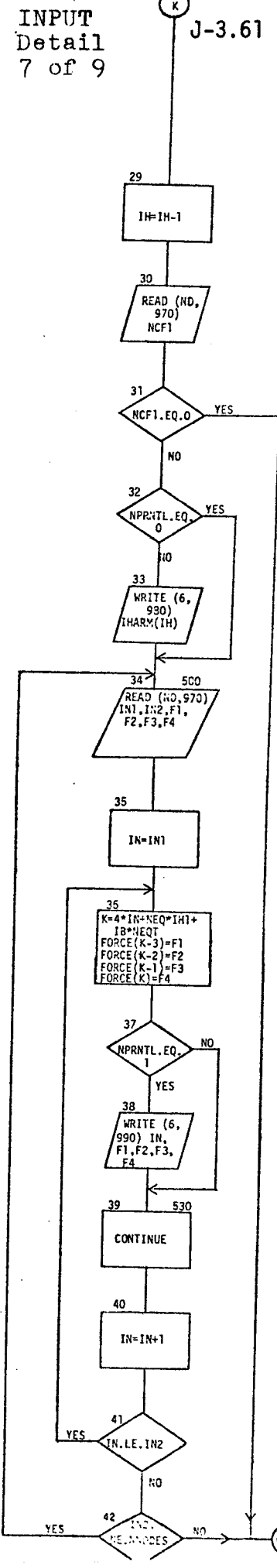
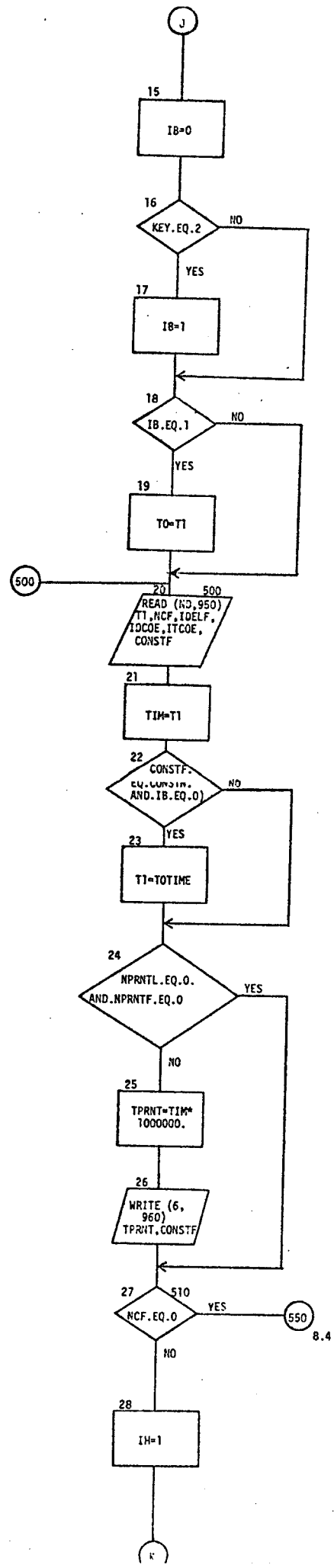
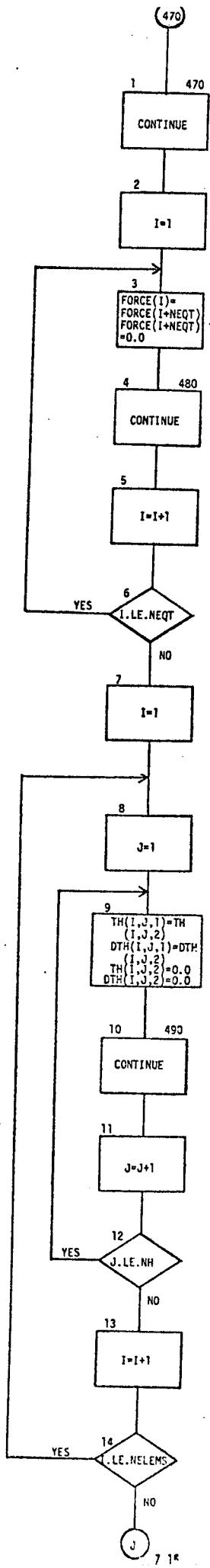
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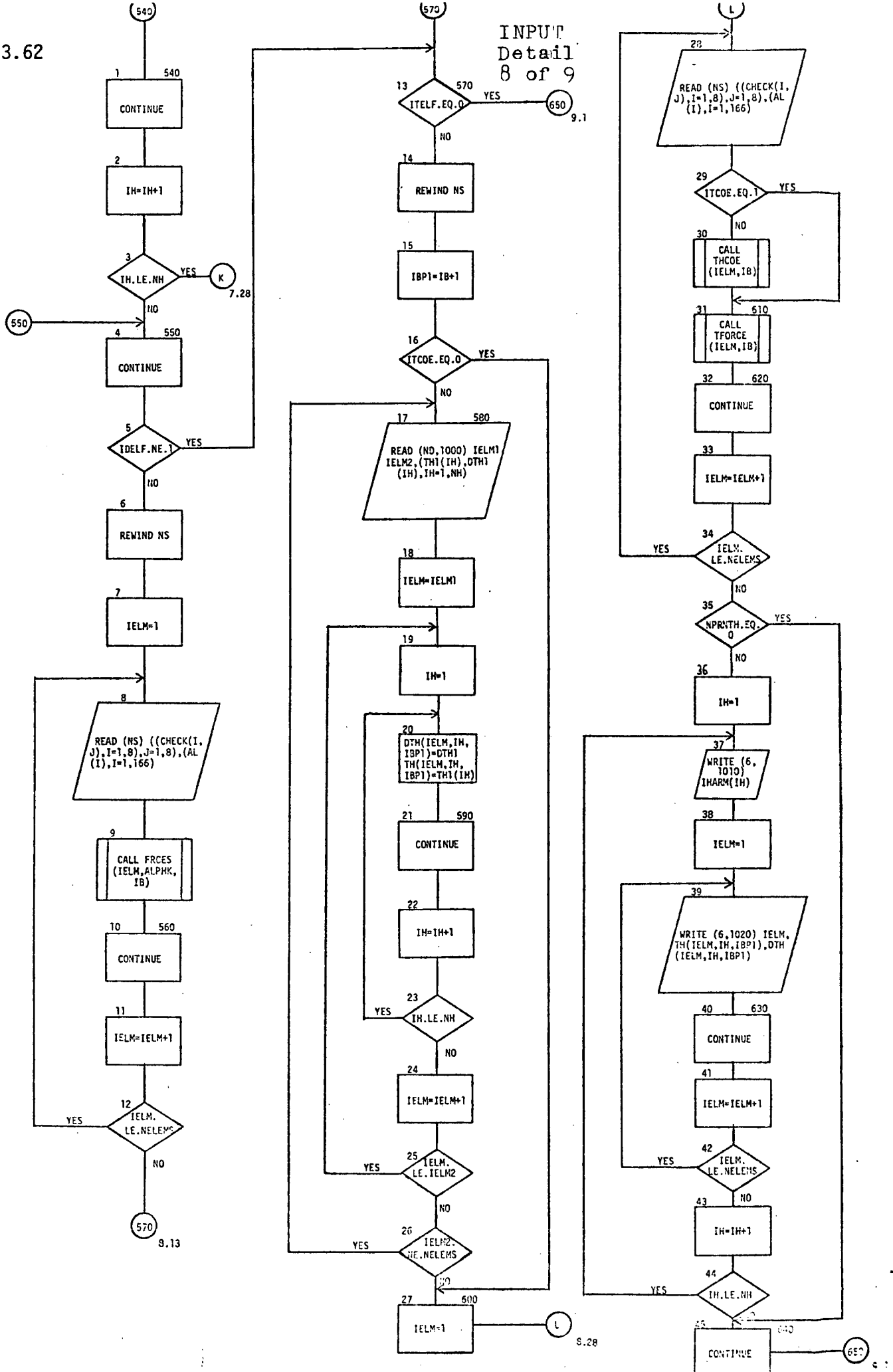
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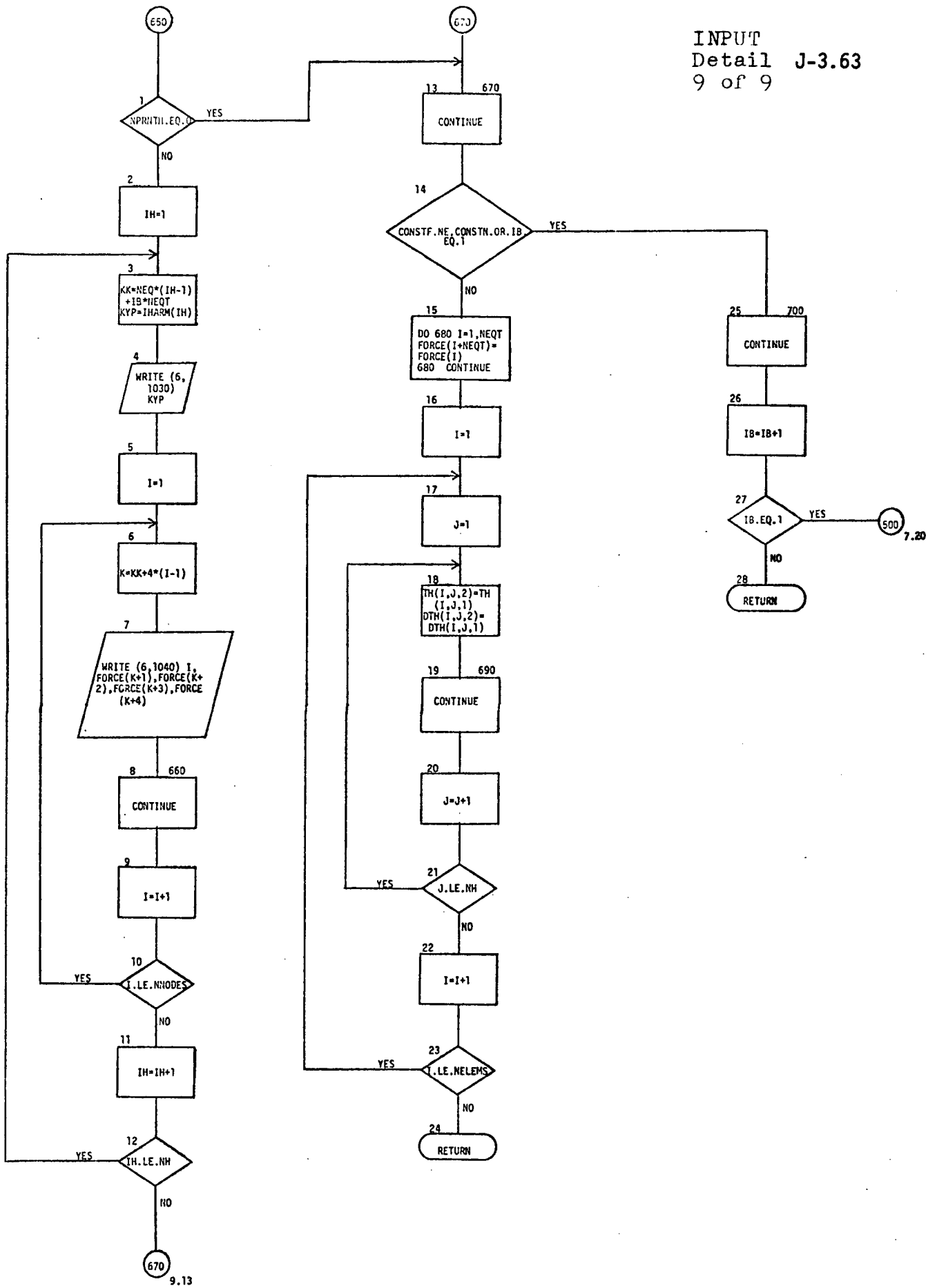
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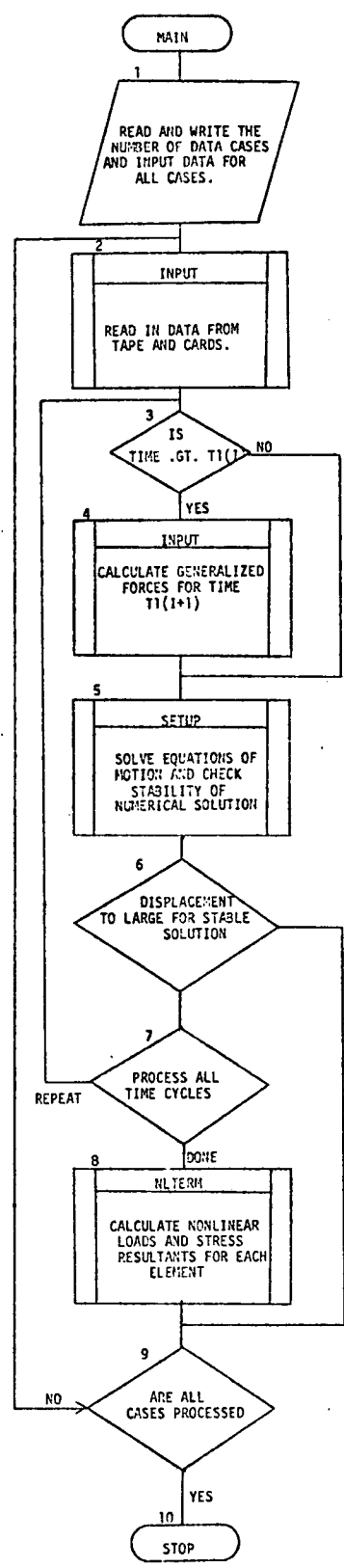


0019		MPRINT=0	DYNC0290
0020	20	READ (5,130) CARD	DYNC0300
0021		IF (CARD(1).EQ.TEST.AND.NCARD.EQ.0) GO TO 40	DYNC0310
0022		NCARD=NCARD+1	DYNC0320
0023		IF (CARD(1).NE.TEST) WRITE (ND,130) CARD	DYNC0330
0024		IF (MPRINT.NE.0) GO TO 30	DYNC0340
0025		WRITE (6,140) NCASE	DYNC0350
0026		WRITE (6,150)	DYNC0360
0027		MPRINT=5000	DYNC0370
0028	30	MPRINT=MPRINT-1	DYNC0380
0029		WRITE (6,160) CARD	DYNC0390
0030		IF (CARD(1).NE.TEST) GO TO 20	DYNC0400
0031		IF (NCARD.NE.1) GO TO 10	DYNC0410
0032	40	NCASE=NCASE-1	DYNC0420
0033		IF (NCASE.NE.NCASES) WRITE (6,170)	DYNC0430
0034		REWIND ND	DYNC0440
0035		NCASE=0	DYNC0460
	C		DYNC0462
0036	50	CALL INPUT (1)	DYNC0470
0037		NCASE=NCASE+1	DYNC0480
0038		LARGE=0	DYNC0490
	C		DYNC0492
	C2	IS THIS A PROGRAM RESTART //YES(70)	DYNC0494
0039		IF (IRSTRT.EQ.1) GO TO 70	DYNC0500
	C		DYNC0508
0040		DO 60 I=1,NEQT	DYNC0510
0041		QP1(I)=0.0	DYNC0520
0042	60	CONTINUE	DYNC0530
	C		DYNC0533
	C2	BEGIN TIME INCREMENTS	DYNC0535
0043		ITP=0	DYNC0550
0044		TIMEP=0.0	DYNC0560
0045		T0=0.0	DYNC0570
0046	70	KKP2=ITP+(TOTIME-TIMEP+DELTE*.001)/DELTE	DYNC0580
0047		NOIT=KKP2	DYNC0590
0048		LL=1	DYNC0600
0049		IF (IRSTRT.EQ.1) LL=ITP+1	DYNC0610
	C1D0	PROCESS ALL TIME CYCLES	DYNC0612
	C		DYNC0618
0050		DO 90 ITAM=LL,KKP2	DYNC0620
0051		TIME=TIMEP+(ITAM-ITP)*DELTE	DYNC0630
	C1IF	TIME .GT. T1(I)//NO(80)	DYNC0632
0052		IF (TIME.LE.T1) GO TO 80	DYNC0640
0053		IF (ITAM.EQ.KKP2) GO TO 80	DYNC0650
0054		T0=TI	DYNC0660
	C1	CALCULATE GENERALIZED FORCES FOR TIME T1(I+1)	DYNC0662
0055		CALL INPUT (2)	DYNC0670
0056	80	CONTINUE	DYNC0680

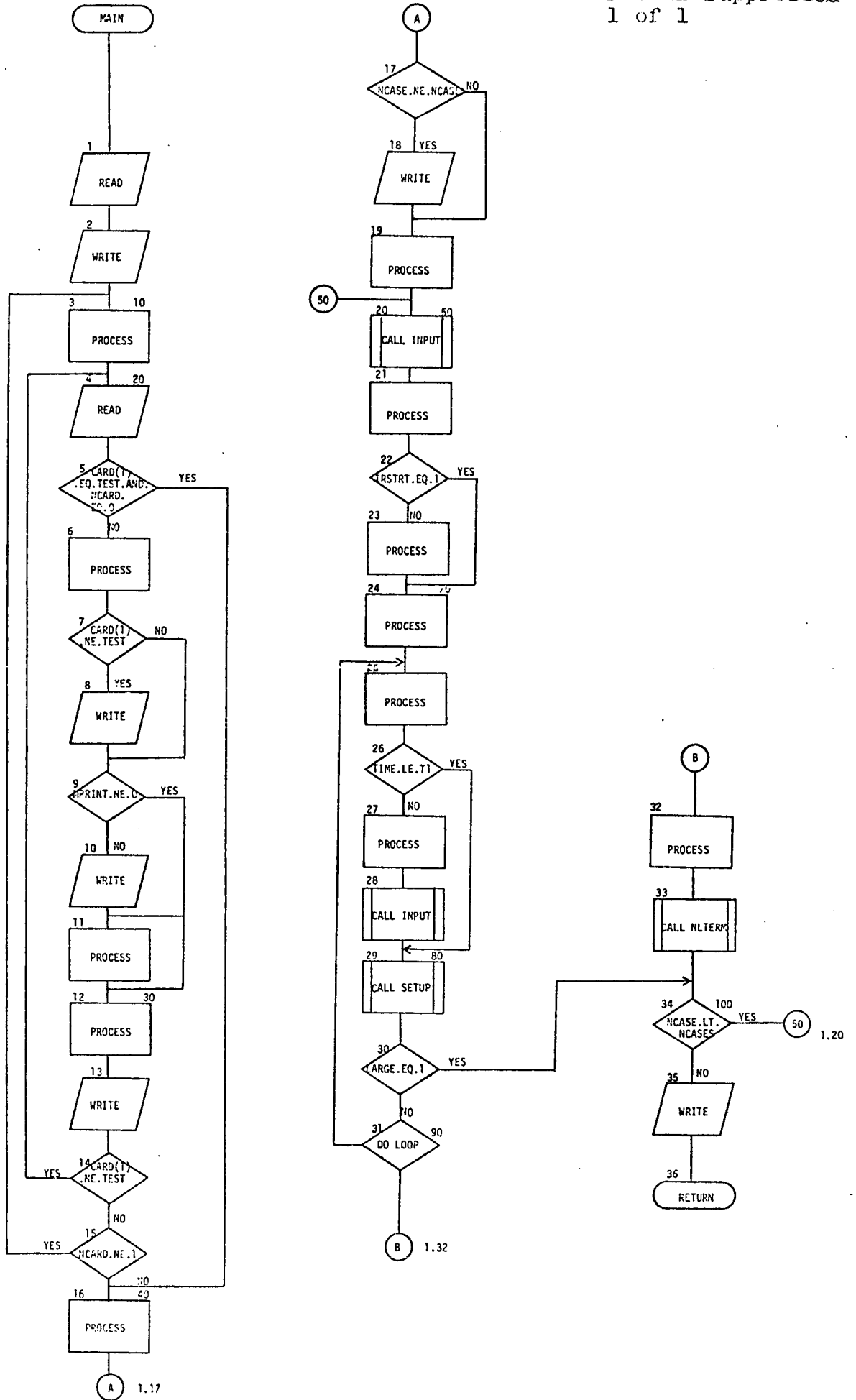
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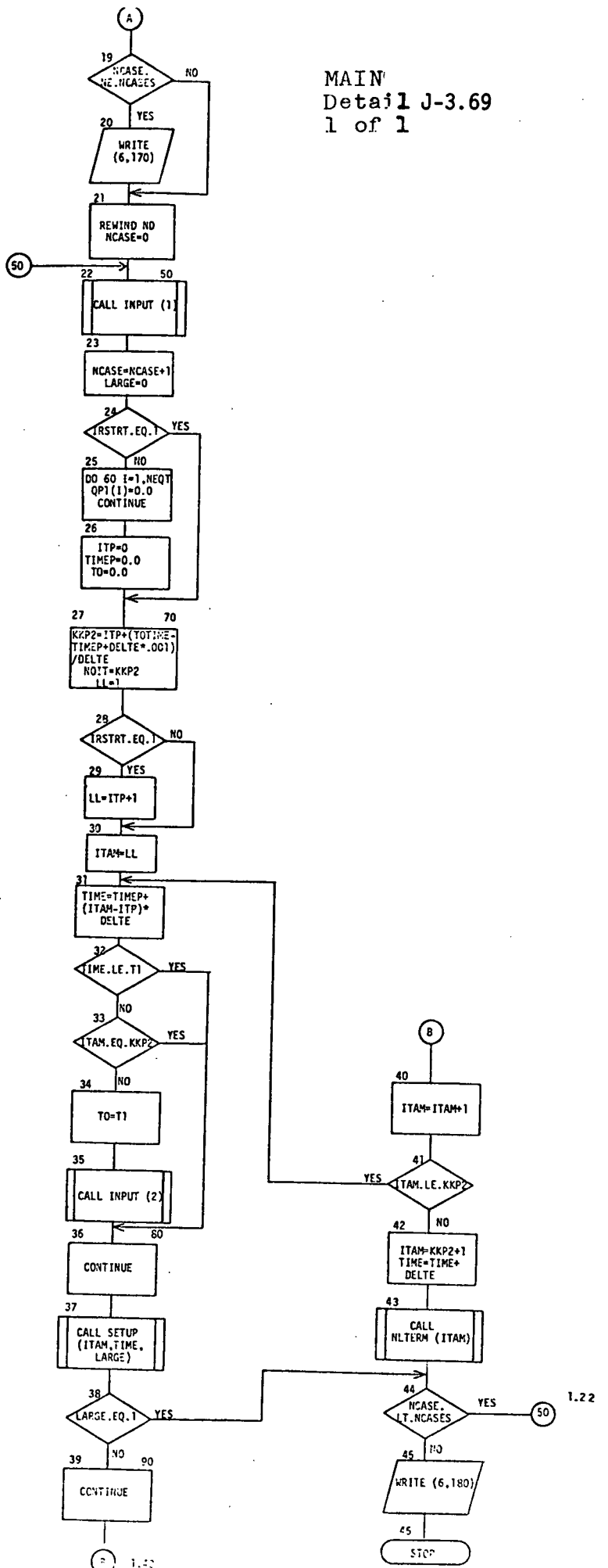
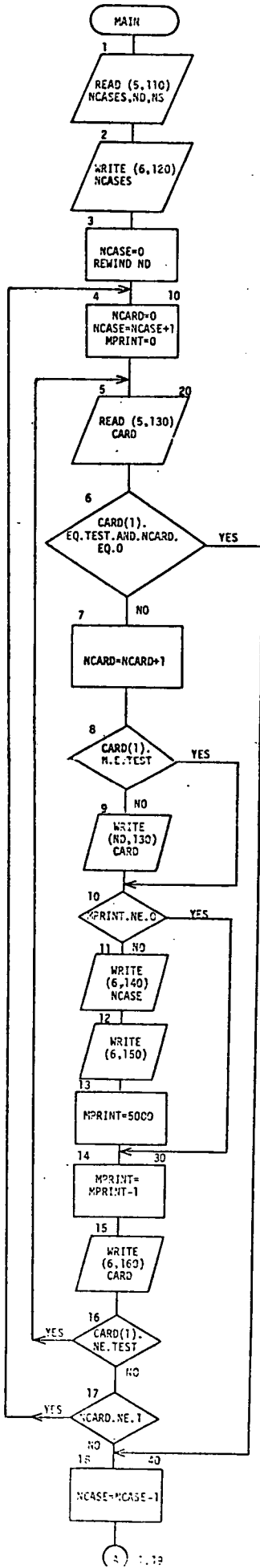
C1      SOLVE EQUATIONS OF MOTION AND CHECK STABILITY OF NUMERICAL      DYN00682
C1C     SOLUTION.                                                       DYN00684
0057    CALL SETUP (ITAM,TIME,LARGE)                                     DYN00690
C1      DISPLACEMENT TOO LARGE FOR STABLE SOLUTION//YES(100)          DYN00692
0058    IF (LARGE.EQ.1) GO TO 100                                       DYN00700
0059    90 CONTINUE                                                       DYN00710
C
0060    ITAM=KKP2+1                                                       DYN00720
0061    TIME=TIME+DELTE                                                  DYN00730
C1      CALCULATE NONLINEAR LOADS AND STRESS RESULTANTS FOR EACH ELEMENT DYN00732
0062    CALL NLTERM (ITAM)                                               DYN00740
C1      ARE ALL CASES PROCESSED//NO(50)                                  DYN00742
0063    100 IF (NCASE.LT.NCASES) GO TO 50                                DYN00750
0064    WRITE (6,180)                                                    DYN00760
0065    STOP                                                             DYN00770
C
0066    110 FORMAT (3I5)                                                 DYN00780
0067    120 FORMAT (1H1,///,30X,31HTHE NUMBER OF CASES TO BE RUN =,I5) DYN00790
0068    130 FORMAT (20A4)                                               DYN00800
0069    140 FORMAT (//8H1 NCASE=,I1//,28X,22HPRINTOUT OF INPUT DATA,/) DYN00820
0070    150 FORMAT (13X,2H10,8X,2H20,8X,2H30,8X,2H40,8X,2H50,8X,2H60,8X,2H70,
      1      8X,2H80/
      1      5X,8(10H1234567890)//)
0071    160 FORMAT (5X,20A4)                                             DYN00860
0072    170 FORMAT (50H THE NUMBER OF INPUT CASES DOES NOT AGREE WITH THE,
      1      22H VALUE OF NCASES INPUT)                                DYN00870
0073    180 FORMAT (1H1//10X,18HALL DATA PROCESSED//10X,11H . . . STOP) DYN00880
0074    END                                                             DYN00900

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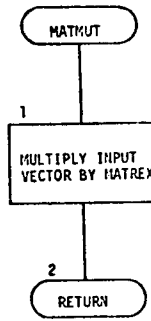


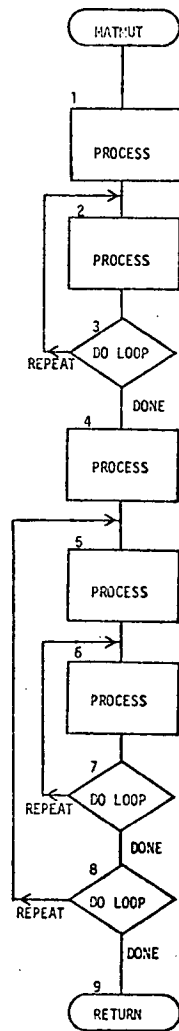


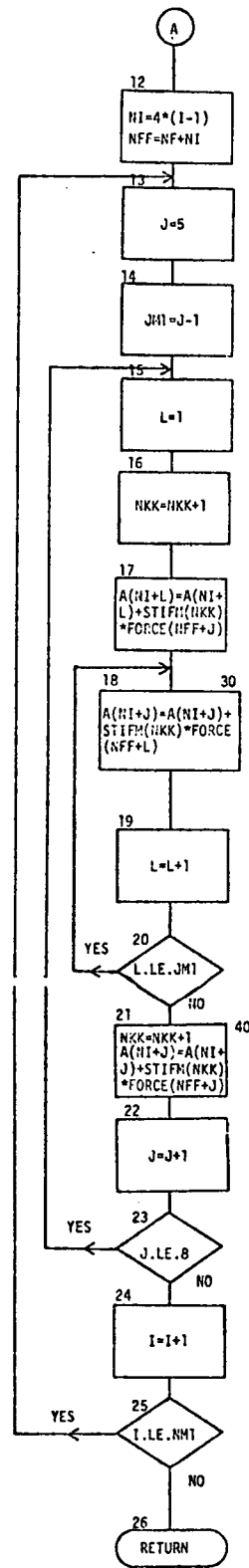
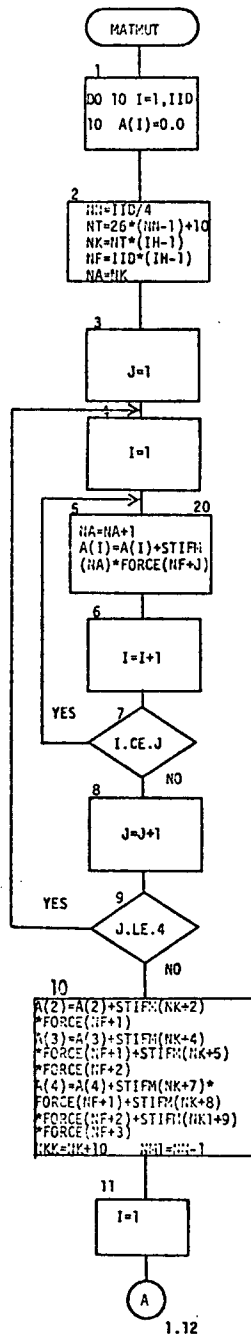
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CE(MATMUT) DYN10852
C DYN10854
C DESCRIPTION - TO MULTIPLY AN INPUT VECTOR BY THE STIFFNESS MATRIX DYN10856
C DYN10858
C INPUT ARGUMENTS. DYN10860
C FORCE = MATRIX OF GENERALIZED FORCES DUE TO EXTERNAL LOADS AND DYN10862
C TEMPERATURES. DYN10864
C IH = HARMONIC KEY. DYN10866
C IID = VECTOR RANK IN MATMUT. DYN10868
C STIFM = STIFFNESS MATRIX. DYN10870
C DYN10872
C EXTERNALS. DYN10874
C CALLED BY DYN10876
C HOUBQ1 DYN10878
C HOUBQN DYN10880
C DYN10882
0001 SUBROUTINE MATMUT (IH,FORCE,STIFM,A,IID) DYN10884
0002 IMPLICIT REAL*8 (A-H,O-Z) DYN10886
0003 DIMENSION A(204), STIFM(6550), FORCE(1020) DYN10888
C1 MULTIPLY INPUT VECTOR BY MATRIX DYN10890
C DYN10938
0004 DO 10 I=1,IID DYN10940
0005 A(I)=0.0 DYN10950
0006 10 CONTINUE DYN10952
C DYN10955
0007 NN=IID/4 DYN10960
0008 NT=26*(NN-1)+10 DYN10970
0009 NK=NT*(IH-1) DYN10980
0010 NF=IID*(IH-1) DYN10990
0011 NA=NK DYN11000
C DYN11008
0012 DO 20 J=1,4 DYN11010
C DYN11018
0013 DO 20 I=1,J DYN11020
0014 NA=NA+1 DYN11030
0015 A(I)=A(I)+STIFM(NA)*FORCE(NF+J) DYN11040
0016 20 CONTINUE DYN11042
C DYN11045
0017 A(2)=A(2)+STIFM(NK+2)*FORCE(NF+1) DYN11050
0018 A(3)=A(3)+STIFM(NK+4)*FORCE(NF+1)+STIFM(NK+5)*FORCE(NF+2) DYN11060
0019 A(4)=A(4)+STIFM(NK+7)*FORCE(NF+1)+STIFM(NK+8)*FORCE(NF+2)+ DYN11070
1 STIFM(NK+9)*FORCE(NF+3) DYN11080
0020 NKK=NK+10 DYN11090
0021 NM1=NN-1 DYN11100
C DYN11108
0022 DO 40 I=1,NM1 DYN11110
0023 NI=4*(I-1) DYN11120
0024 NFF=NF+NI DYN11130
    
```

0025	C		DYN11138
0026		DO 40 J=5,8	DYN11140
		JM1=J-1	DYN11150
0027	C		DYN11158
0028		DO 30 L=1,JM1	DYN11160
0029		NKK=NKK+1	DYN11170
0030		A(NI+L)=A(NI+L)+STIFM(NKK)*FORCE(NFF+J)	DYN11180
0031		A(NI+J)=A(NI+J)+STIFM(NKK)*FORCE(NFF+L)	DYN11190
	30	CONTINUE	DYN11192
0032	C		DYN11195
0033		NKK=NKK+1	DYN11200
0034		A(NI+J)=A(NI+J)+STIFM(NKK)*FORCE(NFF+J)	DYN11210
	40	CONTINUE	DYN11212
0035	C		DYN11215
0036		RETURN	DYN11220
		END	DYN11230







```

C(NLTERM) DYN06532
C DYN06534
C DESCRIPTION - TO PROCESS ALL ELEMENTS TO OBTAIN THE DYN06536
C GENERALIZED NONLINEAR LOADS. IF STRESS COMPUTATIONS DYN06538
C ARE REQUIRED, THEN THE STRESS RESULTANTS AND STRESSES DYN06540
C ON THE UPPER AND LOWER SHELL SURFACES ARE COMPUTED DYN06542
C AND PRINTED OUT. DYN06544
C DYN06546
C INPUT ARGUMENTS. DYN06548
C E1 = MATRIX CONTAINING THE YOUNG'S MODULUS IN THE MERIDIANAL DYN06550
C DIRECTION FOR EACH HARMONIC. DYN06552
C E2 = MATRIX CONTAINING THE YOUNG'S MODULUS IN CIRCUMFERENTIAL DYN06554
C DIRECTION FOR EACH ELEMENT. DYN06556
C FNU1 = MATRIX CONTAINING THE VALUES OF POISSON'S RATIO FOR DYN06558
C EACH ELEMENT. DYN06560
C FNU2 = MATRIX CONTAINING THE VALUES OF POISSON'S RATION FOR EACH DYN06562
C ELEMENT. DYN06564
C G = SHEAR MODULUS, G (FOR AN ISOTROPIC MATERIAL DYN06566
C  $G = (E/2)*(1 + \nu)$ ). DYN06568
C QPR = - PARTIAL DERIVATIVE OF U-NL WITH RESPECT TO LOWER CASE Q DYN06570
C T = MATRIX OF ELEMENT THICKNESSES. DYN06572
C DYN06574
C OUTPUT ARGUMENTS. DYN06576
C QP = Q - PARTIAL DERIVATIVE OF U-NL WITH RESPECT TO LOWER DYN06578
C CASE Q AT TIME STEP (N-1). DYN06580
C DYN06582
C EXTERNALS. DYN06584
C CALLED BY DYN06586
C MAIN DYN06588
C INPUT DYN06590
C SETUP DYN06592
C CALLS DYN06594
C QPRIME DYN06596
C STRESS DYN06598
C DYN06600
0001 SUBROUTINE NLTERM (ITAM) DYN06602
0002 IMPLICIT REAL*8 (A-H,O-Z) DYN06604
0003 COMMON /CONST/ NH,NELEMS,NNODES,NSIZE,NPRNTQ,NEQ,NEQT,N,NN,NHNS, DYN06606
1 DT2,NPRNTL,NPRNTF,IDELF,IDCOE DYN06608
0004 COMMON /TMFT/ TOTIME,DELTE,TIME,T0,T1 DYN06610
0005 COMMON /QS/ QN(1020),QN1(1020),FORCE(2740),QP(1020),QP1(1020), DYN06612
1 QN2(1020) DYN06614
0006 COMMON /GEO/ FNU1(50),FNU2(50),E1(50),E2(50),G(50),T(50), DYN06616
1 SINE(51),COSINE(51),SINM(50),COSM(50),R(50),PH(50), DYN06618
1 PHP(50),ARCL(50) DYN06619
0007 COMMON /EES/ ES(5),ET(5),EST(5),E13(5),E23(5) DYN06620
0008 COMMON /NLTRMS/ QPR(8,5) DYN06630
0009 COMMON /GCD/ CC1,CC2,DD1,DD2,GG1,GG2 DYN06640

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FORTRAN IV G LEVEL 20

NLTERM

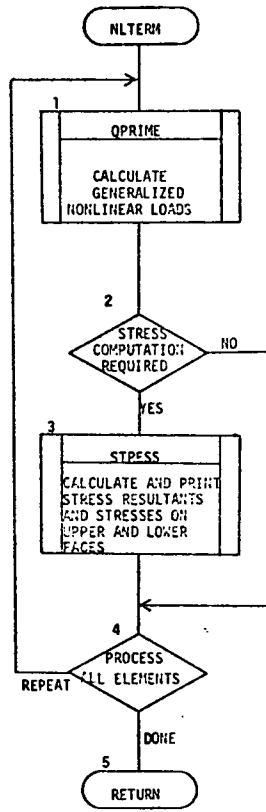
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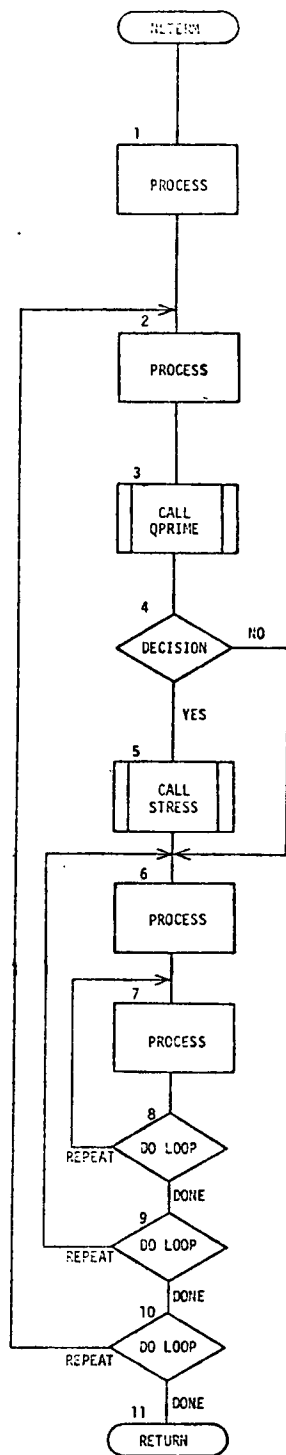
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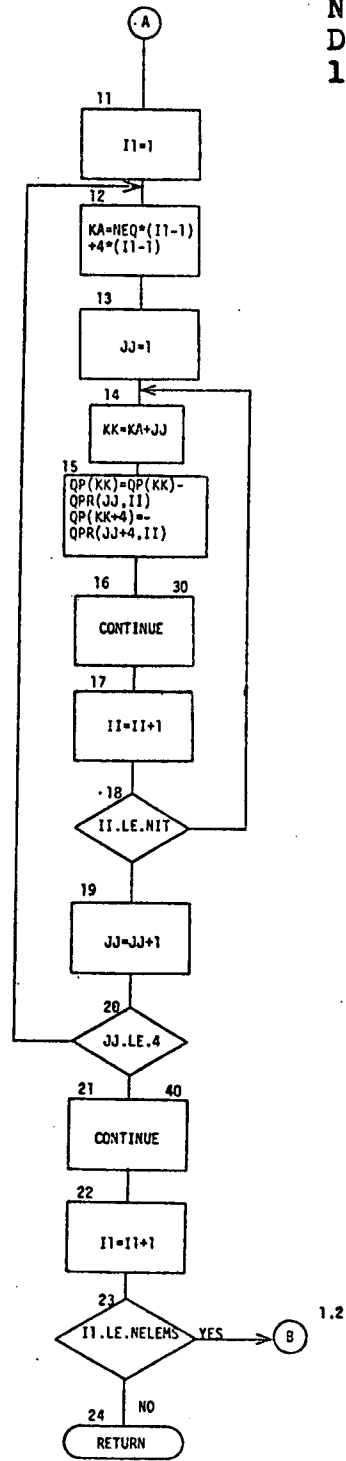
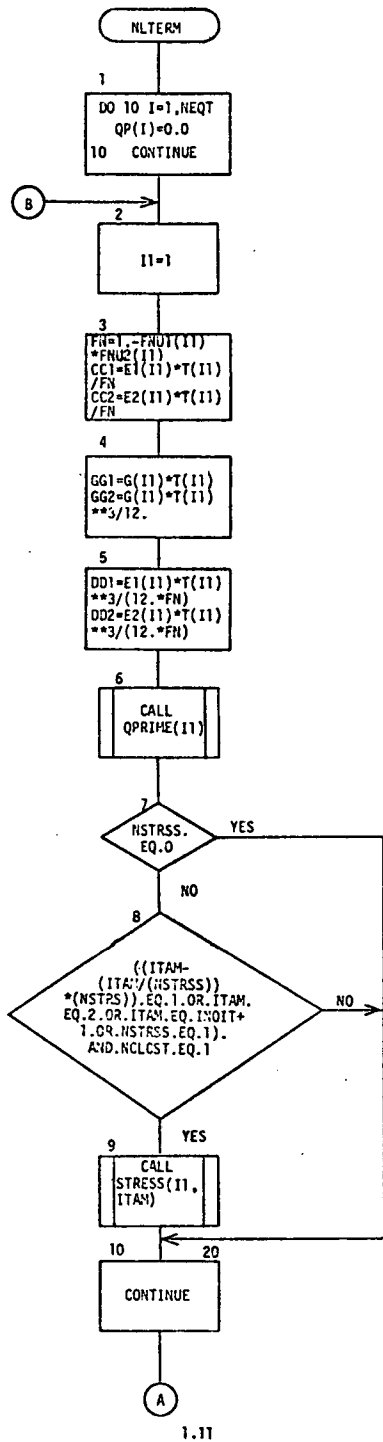
PAGE 0003

```
0032          QP(KK+4)=-QPR(JJ+4,II)
0033      30    CONTINUE
      C
0034      40 CONTINUE
      C
0035          RETURN
0036          END
```

```
DYN07000
DYN07010
DYN07013
DYN07020
DYN07023
DYN07030
DYN07040
```







```

C(NRESTR) DYN10252
C DYN10254
C DESCRIPTION - TO APPLY THE DESIRED BOUNDARY CONDITIONS TO DYN10256
C THE LEFT SIDE OF THE EQUATIONS OF MOTION. INSERT DYN10258
C ONES ON THE DIAGONAL OF THE STIFFNESS MATRIX FOR DYN10260
C THE FIRST HARMONIC ONLY. DYN10262
C DYN10264
C INPUT ARGUMENTS. DYN10266
C KY = RESTART KEY. DYN10268
C LK = MATRIX INDICATING THE NODAL RESTRAINTS WHICH ARE APPLIED DYN10270
C ON THE SHELL. DYN10272
C NCLOSE = CONSTANT USED TO INDICATE THE PRESENCE OF A SINGULARITY DYN10274
C AT THE FIRST NODE OF A CLOSED SHELL. DYN10276
C NN = STORAGE BLOCK INDICATOR FLAG FOR STIFFNESS AND MASS DYN10278
C MATRICES. DYN10280
C NODRES = NUMBER OF DISPLACEMENT CONSTRAINTS APPLIED TO THE SHELL. DYN10282
C DYN10284
C OUTPUT ARGUMENTS. DYN10286
C FORCE = MATRIX OF GENERALIZED FORCES DUE TO EXTERNAL LOADS AND DYN10288
C TEMPERATURES. DYN10290
C STIFM = STIFFNESS MATRIX. DYN10292
C DYN10294
C EXTERNALS. DYN10296
C CALLED BY DYN10298
C HOUBQ1 DYN10300
C DYN10302
C SUBROUTINE NRESTR (KY) DYN10304
C IMPLICIT REAL*8 (A-H,O-Z) DYN10306
C COMMON /SLVEEQ/ STIFM(6550),FORCE(204) DYN10308
C COMMON /CONST/ NH,NELEMS,NNODES,NSIZE,NPRNTQ,NEQ,NEQT,N,NN,NHNS, DYN10310
C 1 DT2,NPRNTL,NPRNTF,IDELEF,IDCOE DYN10312
C COMMON /RSTRNT/ NODRES,NCLOSE,LK(204) DYN10314
C1 APPLY BOUNDARY CONDITIONS TO THE LEFT HAND AND RIGHT HAND SIDES DYN10316
C1C OF THE EQUATIONS OF MOTION DYN10318
C NODRE=NODRES DYN10350
C IF (KY.EQ.0.AND.NCLOSE.EQ.1) NODRE=NODRE+2 DYN10360
C IF (NODRE.EQ.0) GO TO 90 DYN10370
C DYN10378
C DO 80 L=1,NODRE DYN10380
C NEQ=LK(L) DYN10390
C IELM=(NEQ-1)/4 DYN10400
C IF (IELM.GT.0) GO TO 30 DYN10410
C I=NEQ DYN10420
C DYN10428
C DO 10 J=1,NFQ DYN10430
C K=(I*I-1)/2+J+NN DYN10440
C STIFM(K)=0.0 DYN10450
C 10 CONTINUE DYN10452

```

0018	C	J=NEQ	DYN10455
			DYN10460
	C		DYN10468
0019		DO 20 I=NEQ,8	DYN10470
0020		K=(I*I-I)/2+J+NN	DYN10480
0021		STIFM(K)=0.0	DYN10490
0022	20	CONTINUE	DYN10492
	C		DYN10495
0023		K=(NEQ*NEQ-NEQ)/2+NEQ+NN	DYN10500
0024		STIFM(K)=1.0	DYN10510
0025		GO TO 70	DYN10520
0026	30	CONTINUE	DYN10530
0027		I=NEQ-4*IELM	DYN10540
0028		N=I+4	DYN10550
	C		DYN10558
0029		DO 40 J=1,N	DYN10560
0030		K=10+26*(IELM-1)+4*(I-1)+(I*I-I)/2+J+NN	DYN10570
0031		STIFM(K)=0.0	DYN10580
0032	40	CONTINUE	DYN10582
	C		DYN10585
0033		J=NEQ-4*(IELM-1)	DYN10590
0034		N=NEQ-4*IELM	DYN10600
	C		DYN10608
0035		DO 50 I=N,4	DYN10610
0036		K=10+26*(IELM-1)+4*(I-1)+(I*I-I)/2+J+NN	DYN10620
0037		STIFM(K)=0.0	DYN10630
0038	50	CONTINUE	DYN10632
	C		DYN10635
0039		K=10+26*(IELM-1)+4*(N-1)+(N*N-N)/2+J+NN	DYN10640
0040		STIFM(K)=1.0	DYN10650
0041		IF (IELM.EQ.NELEMS) GO TO 70	DYN10660
0042		IELM=IELM+1	DYN10670
0043		J=J-4	DYN10680
	C		DYN10688
0044		DO 60 I=1,4	DYN10690
0045		K=10+26*(IELM-1)+4*(I-1)+(I*I-I)/2+J+NN	DYN10700
0046		STIFM(K)=0.0	DYN10710
0047	60	CONTINUE	DYN10712
	C		DYN10715
0048	70	FORCE(NEQ)=0.0	DYN10720
0049	80	CONTINUE	DYN10730
	C		DYN10733
0050	90	CONTINUE	DYN10740
	C1	IF NOT FIRST HARMONIC .THEN. RETURN	DYN10742
0051		IF (KY.NE.0) RETURN	DYN10780
	C1	INSERT ONES ON THE DIAGONALS OF THE STIFFNESS MATRIX	DYN10782
0052		STIFM(3)=1.0	DYN10790
	C		DYN10798

FORTRAN IV G LEVEL 20

NRESTR

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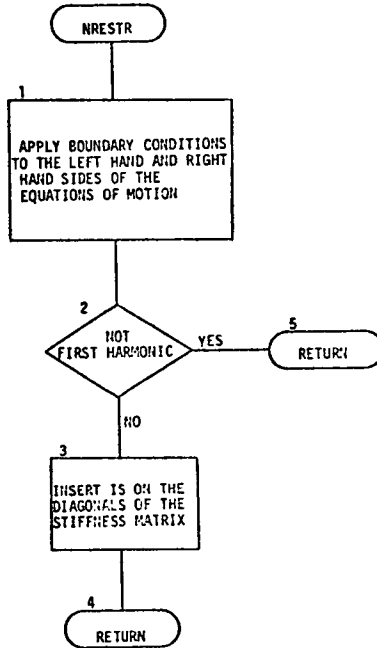
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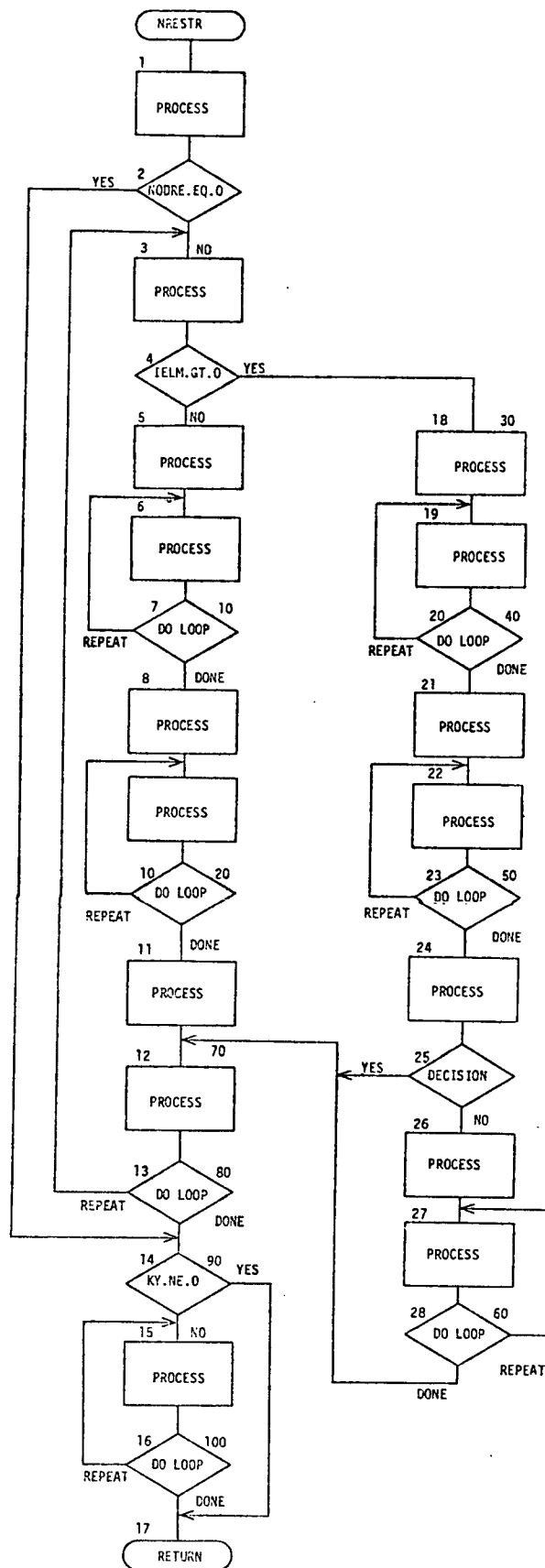
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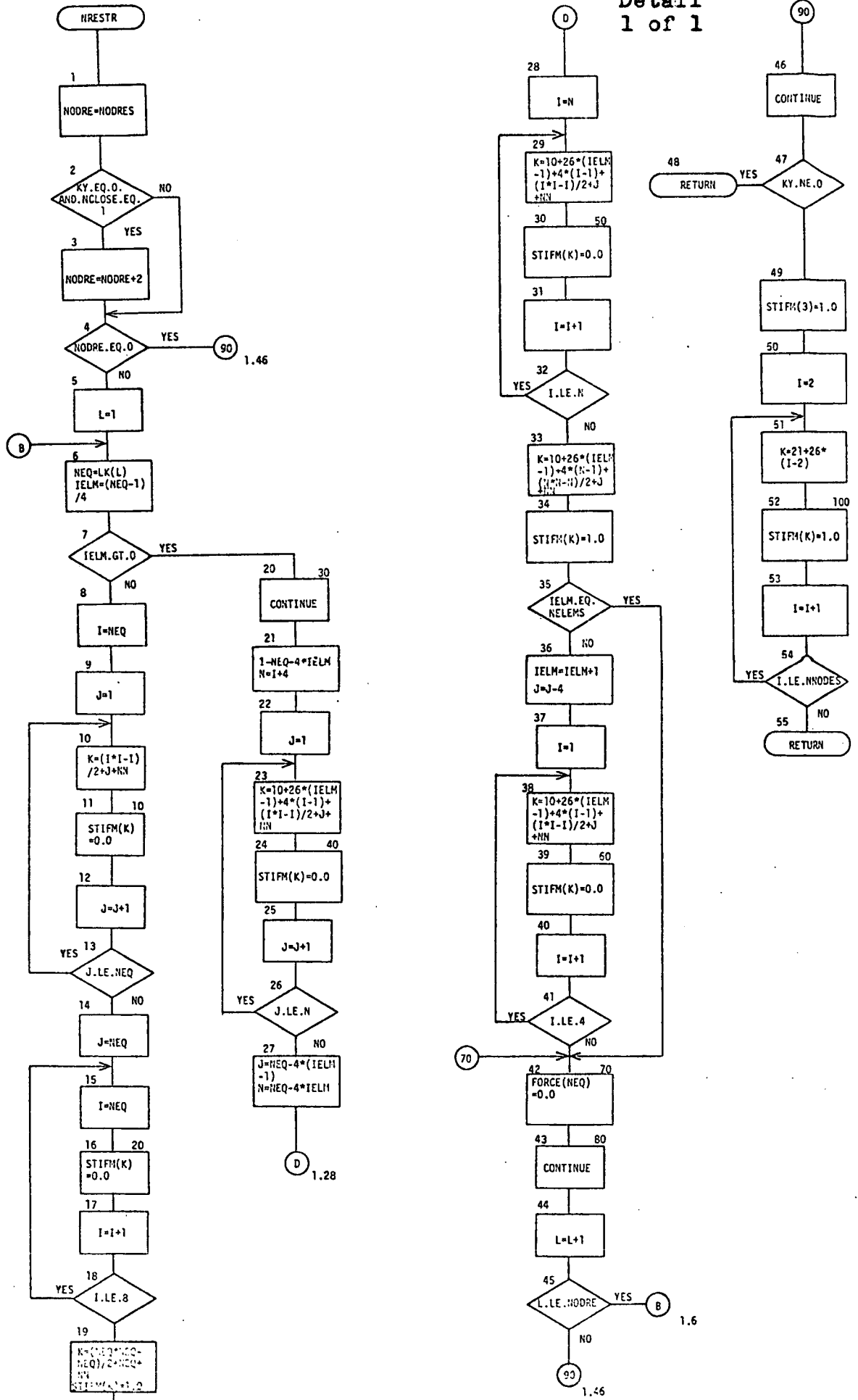
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0053      DO 100 I=2,NNODES
0054          K=21+26*(I-2)
0055          STIFM(K)=1.0
0056      100 CONTINUE
          C
0057      RETURN
0058      END
```

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DYN10800
DYN10810
DYN10820
DYN10822
DYN10825
DYN10830
DYN10840
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C(QPRIME) DYN07052
C DYN07054
C DESCRIPTION - TO COMPUTE THE GENERALIZED NONLINEAR LOADS. DYN07056
C FOURIER COEFFICIENTS AND Q-PRIMES ARE COMPUTED. IF DYN07058
C NECESSARY, COEFFICIENTS AND Q-PRIMES ARE UPDATED DYN07060
C BY INCLUDING THERMAL EFFECTS. DYN07062
C DYN07064
C INPUT ARGUMENTS. DYN07066
C CCC = MATRIX CONTAINING INTEGRALS FROM 0 TO 2*PI OF DYN07068
C COS(I*THETA) * COS(J*THETA) * COS(K*THETA) * DTHETA. DYN07070
C CC1 = YOUNG'S MODULUS TIMES SHELL THICKNESS, NGRMALIZED WITH DYN07072
C RESPECT TO POISSON RATIOS (MERIDIANAL DIRECTION). DYN07074
C CC2 = YOUNG'S MODULUS TIMES SHELL THICKNESS, NORMALIZED WITH DYN07076
C RESPECT TO POISSON RATIOS (CIRCUMFERENTIAL DIRECTION). DYN07078
C CSS = MATRIX CONTAINING INTEGRALS FROM 0 TO 2*PI OF DYN07080
C COS(I*THETA) * SIN(J*THETA) * SIN(K*THETA) * DTHETA. DYN07082
C E13 = MATRIX OF NONLINEAR STRAINS USED IN THE CALCULATION OF DYN07084
C EACH HARMONIC. DYN07086
C E23 = MATRIX OF NONLINEAR STRAINS USED IN THE CALCULATION OF DYN07088
C EACH HARMONIC. DYN07090
C FNU1 = MATRIX CONTAINING THE VALUES OF POISSON'S RATIO FOR DYN07092
C EACH ELEMENT. DYN07094
C GG1 = SHEAR MODULUS TIMES SHELL THICKNESS (MERIDIANAL DYN07096
C DIRECTION). DYN07098
C IHARM = MATRIX OF HARMONIC NUMBERS FOR WHICH DISPLACEMENTS DYN07100
C AND/OR STRESSES WILL BE CALCULATED. DYN07102
C QN = DISPLACEMENTS AT TIME INCREMENT (N-1) UP TO STATEMENT 20 DYN07104
C AFTER STATEMENT 30 THIS MATRIX HAS BEEN CHANGED TO DYN07106
C THE DISPLACEMENTS AT TIME STEP (N). DYN07108
C RO = RADIAL DISTANCE OF ELEMENT FROM ORIGIN. DYN07110
C SSC = MATRIX CONTAINING INTEGRALS FROM 0 TO 2*PI OF DYN07112
C SIN(I*THETA) * SIN(J*THETA) * COS(K*THETA) * DTHETA. DYN07114
C Z = Z-DISTANCE OF ELEMENT FROM ORIGIN. DYN07116
C DYN07118
C OUTPUT ARGUMENTS. DYN07120
C ES = MATRIX OF THE LINEAR STRAINS, USED IN THE CALCULATION DYN07122
C OF EACH HARMONIC. DYN07124
C EST = MATRIX OF THE LINEAR STRAINS, USED IN THE CALCULATION DYN07126
C OF EACH HARMONIC. DYN07128
C ET = MATRIX OF THE LINEAR STRAINS, USED IN THE CALCULATION OF DYN07130
C EACH HARMONIC. DYN07132
C E13 = MATRIX OF NONLINEAR STRAINS USED IN THE CALCULATION OF DYN07134
C EACH HARMONIC. DYN07136
C E23 = MATRIX OF NONLINEAR STRAINS USED IN THE CALCULATION OF DYN07138
C EACH HARMONIC. DYN07140
C QPR = - PARTIAL DERIVATIVE OF U-NL WITH RESPECT TO LOWER CASE Q DYN07142
C DYN07144
C EXTERNALS. DYN07146
    
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C      CALLED BY                                DYN07148
C      NLTERM                                    DYN07150
C
0001      SUBROUTINE QPRIME (I1)                  DYN07152
0002      IMPLICIT REAL*8 (A-H,O-Z)              DYN07154
0003      COMMON /CONST/ NH,NELEMS,NNODES,NSIZE,NPRNTQ,NEQ,NEQT,N,NN,NHNS, DYN07156
1          DT2,NPRNTL,NPRNTF,IDELEF,IDCOE        DYN07158
0004      COMMON /CS/ CCC(125),SSC(125),CSS(125) DYN07160
0005      COMMON /CS4/ CCCC(625),SSSS(625),SSCC(625),SCCS(625) DYN07162
0006      COMMON /EES/ ES(5),ET(5),EST(5),E13(5),E23(5) DYN07164
0007      COMMON /NLTRMS/ QPR(8,5)                DYN07166
0008      COMMON /GEOM/ FNU1(50),FNU2(50),E1(50),E2(50),G(50),T(50), DYN07170
1          SINE(51),COSINE(51),SINM(50),COSM(50),R(50),PH(50), DYN07172
1          PHP(50),ARCL(50)                      DYN07174
0009      COMMON /GCD/ CC1,CC2,DD1,DD2,GG1,GG2   DYN07176
0010      COMMON /HARM/ NHP,IHARM(5)             DYN07178
0011      COMMON /THER/ TH(50,5,2),DTH(50,5,2),ALS(50),ALT(50) DYN07180
0012      COMMON /THCON/ ITSELF,ITCOE,NPRNTH     DYN07182
0013      COMMON /TMFT/ TOTIME,DELTE,TIME,TC,T1  DYN07190
0014      COMMON /QS/ QN(1020),QN1(1020),FORCE(2040),QP(1020),QP1(1020), DYN07200
1          QN2(1020)                              DYN07210
0015      COMMON /RZ/ RO(51),Z(51)               DYN07220
0016      DIMENSION E23Q1(5), E23Q3(5), E23Q5(5), E23Q7(5), ESTQ1(5), DYN07230
1          ESTQ3(5), ESTQ5(5), ESTQ7(5), ETQ2(5), ETQ6(5) DYN07240
C      CALCULATE GENERALIZED NONLINEAR LOADS   DYN07260
C      COMPUTE OFTEN USED QUANTITIES           DYN07280
0017      J1=I1                                  DYN07290
0018      J11=I1+1                               DYN07300
0019      DRO=RO(J11)-RO(J1)                     DYN07310
0020      DZ=Z(J11)-Z(J1)                       DYN07320
0021      ARL=DSQRT(DRO*DRO+DZ*DZ)               DYN07330
0022      SIPH=DRO/ARL                           DYN07340
0023      CUPH=DZ/ARL                            DYN07350
0024      RM=(RO(J11)+RO(J11))/2.0              DYN07360
0025      R2I=1.0/(2.0*RM)                      DYN07370
0026      ARCLI=1.0/ARL                         DYN07380
C      COMPUTE DERIVATES INDEPENDENT OF I     DYN07390
0027      ETQ3=R2I                               DYN07400
0028      ETQ7=R2I                               DYN07410
0029      E23Q2=-R2I*CUPH                       DYN07420
0030      E23Q6=E23Q2                           DYN07430
0031      E13Q1=ARCLI*SIPH                      DYN07440
0032      E13Q3=-ARCLI*CUPH                    DYN07450
0033      E13Q5=-ARCLI*SIPH                    DYN07460
0034      E13Q7=ARCLI*CUPH                     DYN07470
0035      ESTQ2=-SIPH*R2I-ARCLI                 DYN07480
0036      ESTQ6=-SIPH*R2I+ARCLI                 DYN07490
0037      ESQ1=E13Q3                            DYN07500
    
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0038      ESQ3=-E13Q1      DYNC7510
0039      ESQ5=E13Q7      DYN07520
0040      ESQ7=-E13Q5      DYN07530
C1      COMPUTE DERIVATES THAT ARE A FUNCTION OF I      DYNC7532
0041      CO2R=COPH*R2I      DYN07550
0042      SL2R=SIPH*R2I      DYNC7560
0043      CL2R=COPH*R2I      DYN07570
0044      SO2R=SIPH*R2I      DYN07580
C      DYN07588
0045      DO 10 IH=1,NH      DYN07590
0046          K=IHARM(IH)      DYN07600
0047          XK=K      DYNC7610
0048          E23Q1(IH)=SO2R*XK      DYNC7620
0049          E23Q3(IH)=-CO2R*XK      DYN07630
0050          E23Q5(IH)=SL2R*XK      DYN07640
0051          E23Q7(IH)=-CL2R*XK      DYN07650
0052          ESTQ1(IH)=E23Q3(IH)      DYN07660
0053          ESTQ3(IH)=-E23Q1(IH)      DYNC7670
0054          ESTQ5(IH)=E23Q7(IH)      DYN07680
0055          ESTQ7(IH)=-E23Q5(IH)      DYN07690
0056          ETQ2(IH)=R2I*XK      DYN07700
0057          ETQ6(IH)=ETQ2(IH)      DYN07710
C      COMPUTE ET, ES, EST, E13, E23      DYNC7720
0058          KK=NEQ*(IH-1)+4*(I1-1)      DYN07730
0059          KK1=KK+1      DYN07740
0060          KK2=KK+2      DYN07750
0061          KK3=KK+3      DYNC7760
0062          KK5=KK+5      DYNC7770
0063          KK6=KK+6      DYN07780
0064          KK7=KK+7      DYN07790
0065          ET(IH)=+ETQ2(IH)*QN(KK2)+ETQ3*QN(KK3)+ETQ6(IH)*QN(KK6)+ETQ7*
1              QN(KK7)      DYNC7810
0066          ES(IH)=ESQ1*QN(KK1)+ESQ3*QN(KK3)+ESQ5*QN(KK5)+ESQ7*QN(KK7)      DYN07820
0067          EST(IH)=ESTQ1(IH)*QN(KK1)+ESTQ2*QN(KK2)+ESTQ3(IH)*QN(KK3)+
1              ESTQ5(IH)*QN(KK5)+ESTQ6*QN(KK6)+ESTQ7(IH)*QN(KK7)      DYN07840
0068          E13(IH)=E13Q1*QN(KK1)+E13Q3*QN(KK3)+E13Q5*QN(KK5)+E13Q7*QN(KK7)      DYN07850
0069          E23(IH)=E23Q1(IH)*QN(KK1)+E23Q2*QN(KK2)+E23Q3(IH)*QN(KK3)+
1              E23Q5(IH)*QN(KK5)+E23Q6*QN(KK6)+E23Q7(IH)*QN(KK7)      DYN07860
0070      10 CONTINUE      DYN07880
C      DYN07883
0071          ITH=0      DYN07890
0072          RSL=RM*ARL/2.0      DYNC7900
C1      COMPUTE COEFFICIENTS AND Q-PRIMES      DYN07902
C      DYN07908
0073      DO 40 M=1,NH      DYNC7910
0074          CES=0.0      DYN07920
0075          CET=0.0      DYNC7930
0076          CEST=0.0      DYN07940

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0077          CE13=0.0          DYN07950
0078          CE23=0.0          DYN07960
0079          K=IHARM(M)        DYN07970
                                DYN07978
C                                DYN07980
0080          DO 20 I=1,8      DYN07990
0081             QPR(I,M)=0.0   DYN08000
0082          CONTINUE        DYN08003
                                DYN08008
C                                DYN08010
C                                DYN08020
0083          DO 30 I=1,NH     DYN08028
0084             II=IHARM(I)    DYN08030
                                DYN08040
C                                DYN08050
0085          DO 30 J=1,NH     DYN08060
0086             JJ=IHARM(J)    DYN08070
0087             IPJ=II+JJ      DYN08080
0088             IMJ=II-JJ      DYN08082
0089             IF (IPJ.NE.K.AND.IABS(IMJ).NE.K) GO TO 30
0090             ITH=ITH+1      DYN08100
                                DYN08110
C                                DYN08120
0091          COMPUTE COEFFICIENTS
0092             CES=CC1*(CCC(ITH)*E13(I)*E13(J)+FNU1(I1)*SSC(ITH)*E23(I)*
1                 E23(J))+CFS   DYN08130
0093             CET=SSC(ITH)*CC2*E23(I)*E23(J)+FNU1(I1)*CC1*CCC(ITH)*
1                 E13(I)*E13(J)+CET DYN08140
0094             CEST=2.0*GG1*CSS(ITH)*E13(J)*E23(I)+CEST DYN08150
0095             CE13=2.0*CC1*CCC(ITH)*E13(I)*(ES(J)+FNU1(I1)*ET(J))+2.0*
1                 GG1*SSC(ITH)*EST(I)*E23(J)+CE13 DYN08170
0096             CE23=2.0*CSS(ITH)*(E23(I)*(CC2*ET(J)+FNU1(I1)*CC1*ES(J))+
1                 GG1*E13(J)*EST(I))+CE23 DYN08180
0096          CONTINUE        DYN08190
                                DYN08193
C                                DYN08200
C                                DYN08210
0097          COMPUTE Q PRIMES
0098             QPR(1,M)=(CE13*E13Q1+CES*ESQ1+CEST*ESTQ1(M)+CE23*E23Q1(M))*RSL+
1                 QPR(1,M)      DYN08220
0099             QPR(2,M)=(CE23*E23Q2+CEST*ESTQ2+CET*ETQ2(M))*RSL+QPR(2,M) DYN08230
0100             QPR(3,M)=(CE13*E13Q3+CET*ETQ3+CES*ESQ3+CE23*E23Q3(M)+CEST*
1                 ESTQ3(M))*RSL+QPR(3,M) DYN08240
0101             QPR(5,M)=(CE13*E13Q5+CES*ESQ5+CE23*E23Q5(M)+CEST*ESTQ5(M))*RSL+
1                 QPR(5,M)      DYN08260
0102             QPR(6,M)=(CET*ETQ6(M)+CE23*E23Q6+CEST*ESTQ6)*RSL+QPR(6,M) DYN08270
0103             QPR(7,M)=(CE13*E13Q7+CET*ETQ7+CES*ESQ7+CE23*E23Q7(M)+CEST*
1                 ESTQ7(M))*RSL+QPR(7,M) DYN08280
0103          CONTINUE        DYN08290
                                DYN08300
C                                DYN08310
C                                DYN08313
0104          IFO=0           DYN08320
                                DYN08328
C                                DYN08330
0105          DO 60 L=1,NH     DYN08330
0106             CE413=0.0      DYN08340
0107             CE423=0.0      DYN08350

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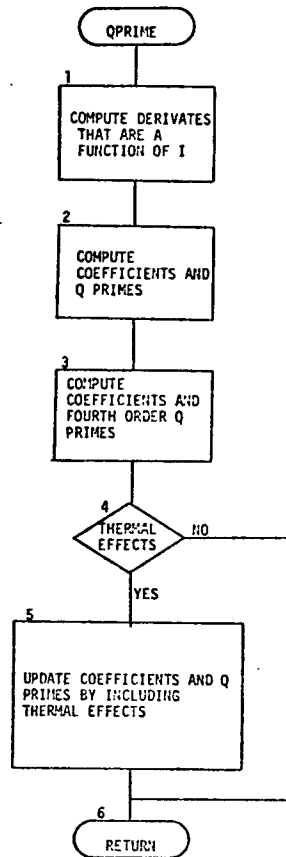
0108      LL=IHARM(L)                                DYN08360
          C                                           DYN08368
0109      DO 50 I=1,NH                                DYN08370
0110      II=IHARM(I)                                  DYN08380
          C                                           DYN08388
0111      DO 50 J=1,NH                                DYN08390
0112      JJ=IHARM(J)                                  DYN08400
0113      IPJ=II+JJ                                    DYN08410
0114      IMJ=IABS(II-JJ)                              DYN08420
          C                                           DYN08428
0115      DO 50 K=1,NH                                DYN08430
0116      KK=IHARM(K)                                  DYN08440
0117      KPL=KK+LL                                    DYN08450
0118      KML=IABS(KK-LL)                              DYN08460
0119      IF (IPJ.NE.KPL.AND.IPJ.NE.KML.AND.IMJ.
          1      NE.KPL.AND.IMJ.NE.KML) GO TO 50      DYN08470
0120      IFO=IFO+1                                    DYN08480
          C1                                           DYN08490
0121      COMPUTE COEFFICIENTS AND FOURTH ORDER Q-PRIMES DYN08492
0122      FOR=(FNU1(II)*CC1+2.0*GG1)*E23(I)*E13(K)    DYN08510
          1      CE413=CC1*E13(I)*E13(J)*E13(K)*CCCC(IFO)+FOR*E23(J)* DYN08520
          SSSC(IFO)+CE413                                DYN08530
0123      CE423=CC2*E23(I)*E23(J)*E23(K)*SSSS(IFO)+FOR*E13(J)* DYN08540
          1      SCCS(IFO)+CE423                          DYN08550
0124      50 CONTINUE                                  DYN08560
          C                                           DYN08563
0125      QPR(1,L)=QPR(1,L)+RSL*(CE413*E13Q1+CE423*E23Q1(L)) DYN08580
0126      QPR(2,L)=QPR(2,L)+RSL*(CE423*E23Q2)         DYN08590
0127      QPR(3,L)=QPR(3,L)+RSL*(CE413*F13Q3+CE423*E23Q3(L)) DYN08600
0128      QPR(5,L)=QPR(5,L)+RSL*(CE413*E13Q5+CE423*E23Q5(L)) DYN08610
0129      QPR(6,L)=QPR(6,L)+RSL*(CE423*E23Q6)         DYN08620
0130      QPR(7,L)=QPR(7,L)+RSL*(CE413*E13Q7+CE423*E23Q7(L)) DYN08630
0131      60 CONTINUE                                  DYN08640
          C                                           DYN08643
          C1 THERMAL EFFECTS//NO(90)                   DYN08645
0132      IF (ITELF.EQ.0) GO TO 90                     DYN08650
          C1 UPDATE COEFFICIENTS AND Q-PRIMES BY INCLUDING THERMAL DFFECTS DYN08652
0133      ITH=0                                         DYN08660
0134      RL=RM*ARL                                     DYN08670
          C                                           DYN08678
0135      DO 80 M=1,NH                                  DYN08680
0136      CE13=0.0                                       DYN08690
0137      CE23=0.0                                       DYN08700
0138      K=IHARM(M)                                    DYN08710
          C                                           DYN08718
0139      DO 70 I=1,NH                                  DYN08720
0140      II=IHARM(I)                                    DYN08730
          C                                           DYN08738
0141      DO 70 J=1,NH                                  DYN08740

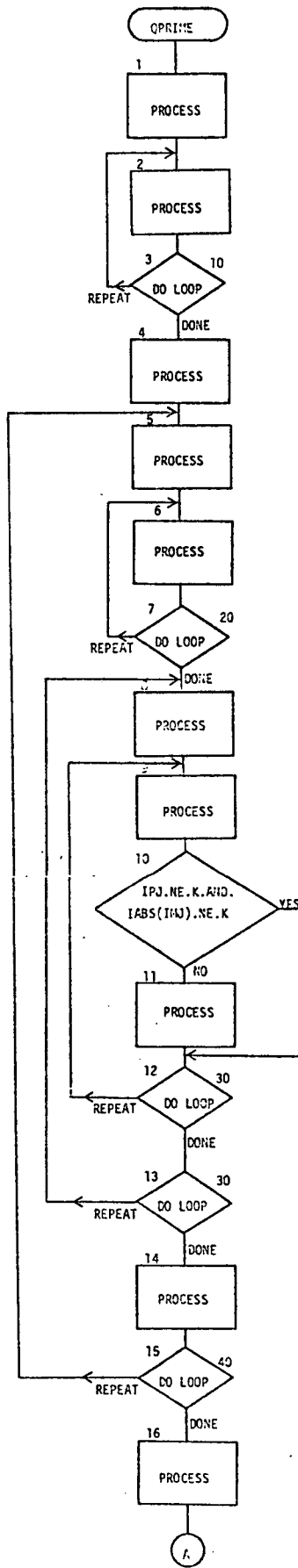
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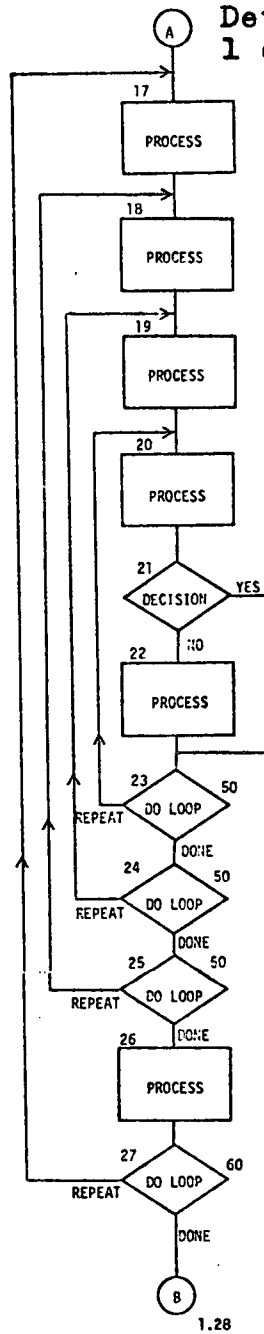
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0142          JJ=IHARM(J)                                DYN08750
0143          IPJ=II+JJ                                  DYN08760
0144          IMJ=II-JJ                                  DYN08770
0145          IF (IPJ.NE.K.AND.IABS(IMJ).NE.K) GO TO 70  DYN08780
0146          THT=TH(I1,J,1)+(TH(I1,J,2)-TH(I1,J,1))*(TIME-TC)/(T1-T0) DYN08790
0147          ITH=ITH+1                                    DYN08800
              C                                         DYN08810
              COMPUTE COEFFICIENTS
0148          CE13=-CC1*(ALS(I1)+FNU1(I1)*ALT(I1))*THT*CCC(ITH)*E13(I)+DYN08820
              1          CE13                                DYN08822
0149          CE23=-CC2*(ALT(I1)+FNU2(I1)*ALS(I1))*THT*CSS(ITH)*E23(I)+DYN08830
              1          CE23                                DYN08832
0150          70 CONTINUE                                DYN08840
              C                                         DYN08843
              C                                         DYN08850
              COMPUTE Q PRIMES
0151          QPR(1,M)=QPR(1,M)+(CE13*E13Q1+CE23*E23Q1(M))*RL DYN08860
0152          QPR(2,M)=QPR(2,M)+CE23*E23Q2*RL          DYN08870
0153          QPR(3,M)=QPR(3,M)+(CE13*E13Q3+CE23*E23Q3(M))*RL DYN08880
0154          QPR(5,M)=QPR(5,M)+(CE13*E13Q5+CE23*E23Q5(M))*RL DYN08890
0155          QPR(6,M)=QPR(6,M)+CE23*E23Q6*RL          DYN08900
0156          QPR(7,M)=QPR(7,M)+(CE13*E13Q7+CE23*E23Q7(M))*RL DYN08910
0157          80 CONTINUE                                DYN08920
              C                                         DYN08923
              C                                         DYN08930
0158          90 CONTINUE                                DYN08930
0159          RETURN                                     DYN08940
0160          END                                         DYN08950
    
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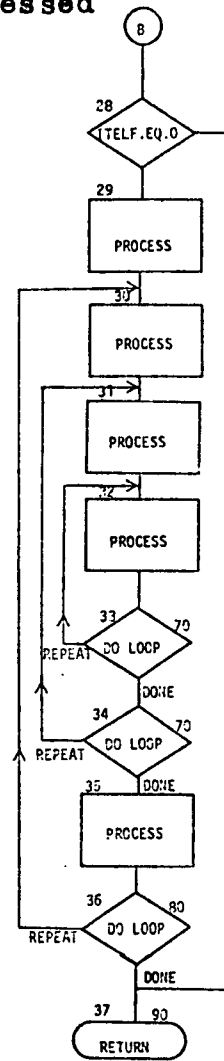




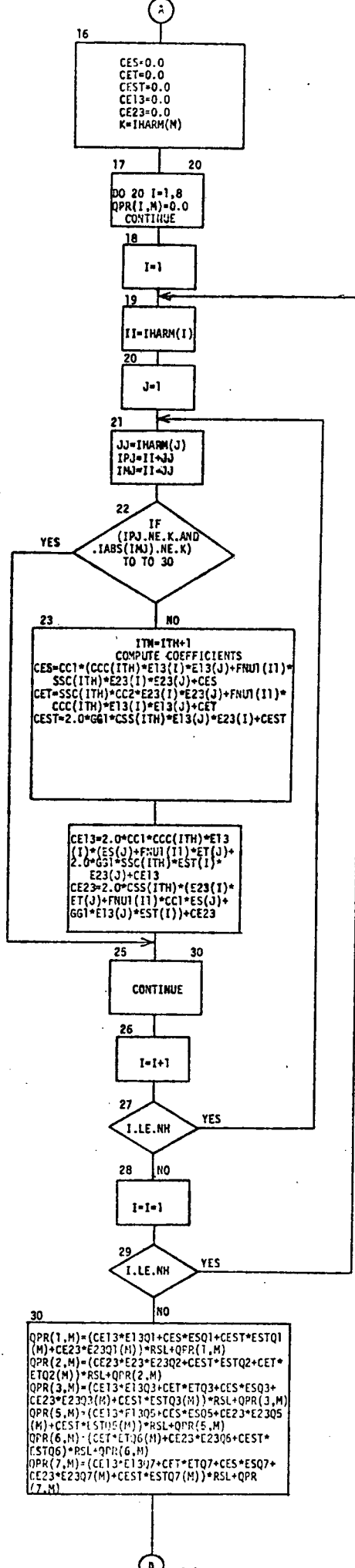
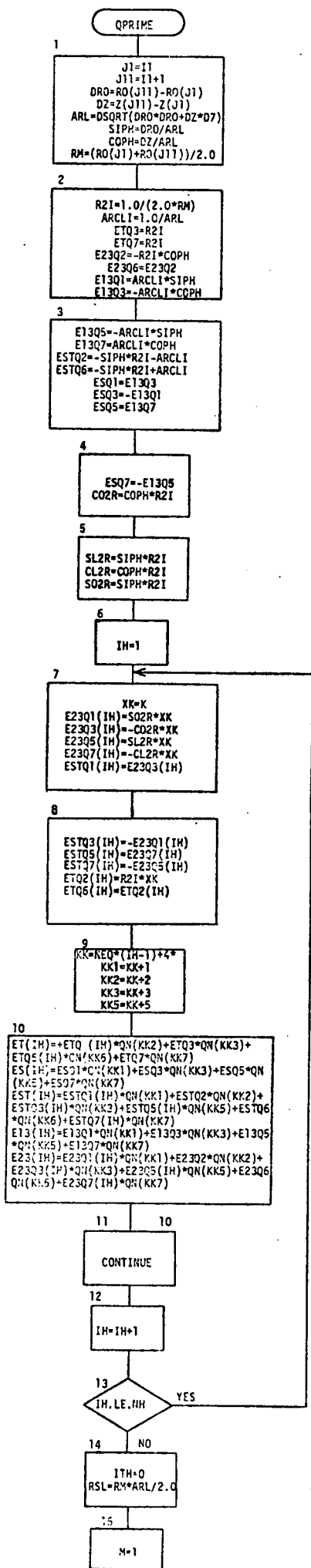
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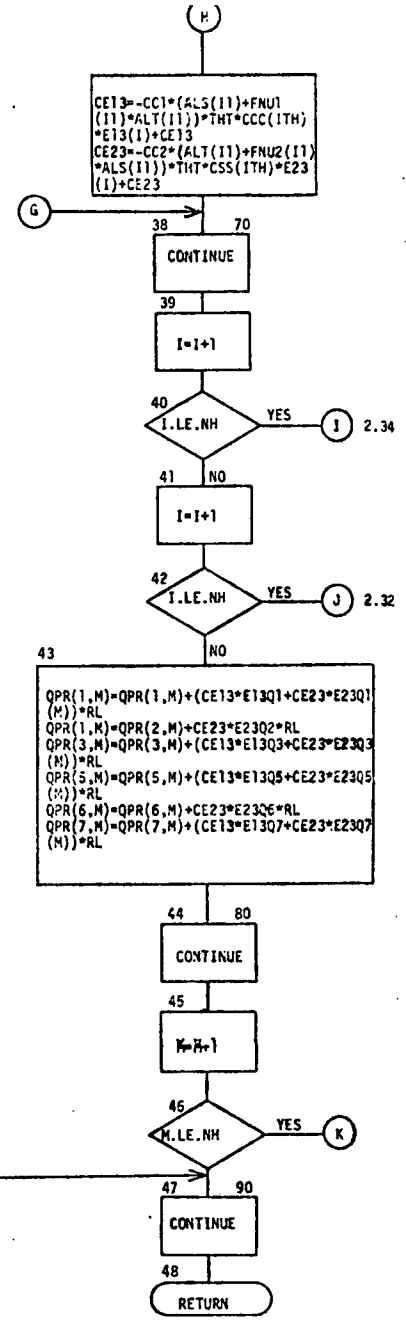
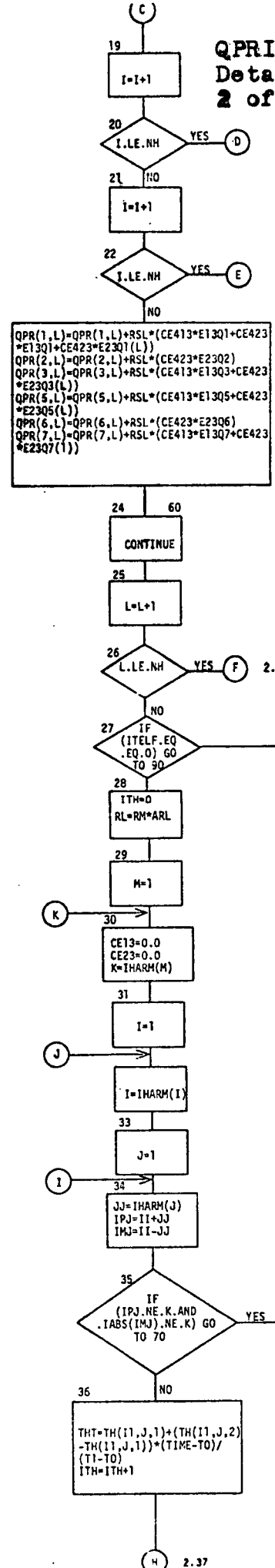
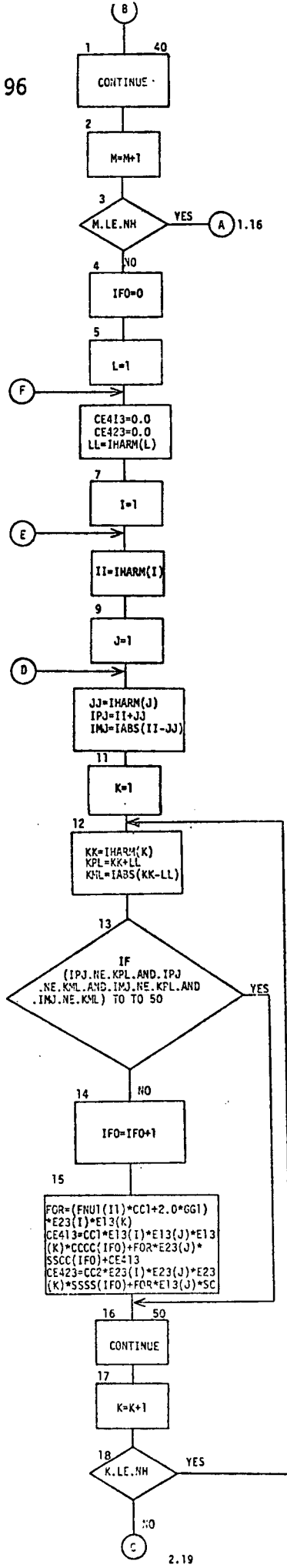


1.28



1.29





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CE(SETUP) DYN05662
C DYN05664
C DESCRIPTION - TO OBTAIN THE SHELL DISPLACEMENTS BY SOLVING DYN05666
C THE EQUATIONS OF MOTION. THESE DISPLACEMENTS ARE DYN05668
C CHECKED IN MAGNITUDE FOR BEING TOO LARGE TO ALLOW DYN05670
C STABLE SOLUTIONS. THE AXIAL DISPLACEMENT OF THE SECOND DYN05672
C NODE IS CHECKED. DYN05674
C DYN05676
C INPUT ARGUMENTS. DYN05678
C NH = TOTAL NUMBER OF HARMONICS USED IN THE DYNAMIC ANALYSIS. DYN05680
C QN = DISPLACEMENTS AT TIME INCREMENT (N-1) UP TO STATEMENT 20 DYN05682
C AFTER STATEMENT 30 THIS MATRIX HAS BEEN CHANGED TO DYN05684
C THE DISPLACEMENTS AT TIME STEP (N). DYN05686
C THETA = MATRIX CONTAINING CIRCUMFERENTIAL ANGLES AT WHICH DYN05688
C STRESSES AND/OR DISPLACEMENTS ARE TO BE CALCULATED. DYN05690
C DYN05692
C OUTPUT ARGUMENTS. DYN05694
C LARGE = CONSTANT WHICH CONTROLS TERMINATION OF THE PROBLEM DYN05696
C IF DISPLACEMENTS BECOME EXCESSIVE. DYN05698
C QLOAD = RIGHT-HAND SIDE OF THE DYNAMIC EQUATIONS OF MOTION DYN05700
C BEFORE CALLING SOLVEQ. DYN05702
C DYN05704
C EXTERNALS. DYN05706
C CALLED BY DYN05708
C MAIN DYN05710
C CALLS DYN05712
C NLTERM DYN05714
C HOUBQN DYN05716
C HOUBQ1 DYN05718
C INPUT DYN05720
C DYN05722
0001 SUBROUTINE SETUP (ITAM,TIME,LARGE) DYN05724
0002 IMPLICIT REAL*8 (A-H,O-Z) DYN05726
0003 COMMON /SLVEEQ/ XN(6550),QLOAD(204) DYN05728
0004 COMMON /QS/ QN(1020),QN1(1020),FORCE(2040),QP(1020),QP1(1020), DYN05730
1 QN2(1020) DYN05732
0005 COMMON /CONST/ NH,NELEMS,NNODES,NSIZE,NPRNTQ,NEQ,NEQT,N,NN,NHNS, DYN05734
1 DT2,NPRNTL,NPRNTF,IDELF,IDCOE DYN05736
0006 COMMON /GCD/ CC1,CC2,DD1,DD2,GG1,GG2 DYN05738
0007 COMMON /THETAS/ THETA(20),NTHETA,NCLCST,NSTRSS DYN05740
0008 COMMON /PRINT/ IPRINT,NOIT,LL DYN05750
0009 COMMON /HARM/ NHP,IHARM(5) DYN05760
0010 COMMON /RESTR/ IRSTRT,NPRNT,NPRNIT,ITP,TIMEP,DELTEP DYN05770
C1IF NOT FIRST POINT .THEN. CALCULATE GENERALIZED NONLINEAR LOADS AND DYN05772
C1C STRESS RESULTANTS DYN05774
0011 IF (ITAM.NE.LL) CALL NLTERM (ITAM) DYN05820
C1DO PROCESS HARMONICS DYN05822
C DYN05828

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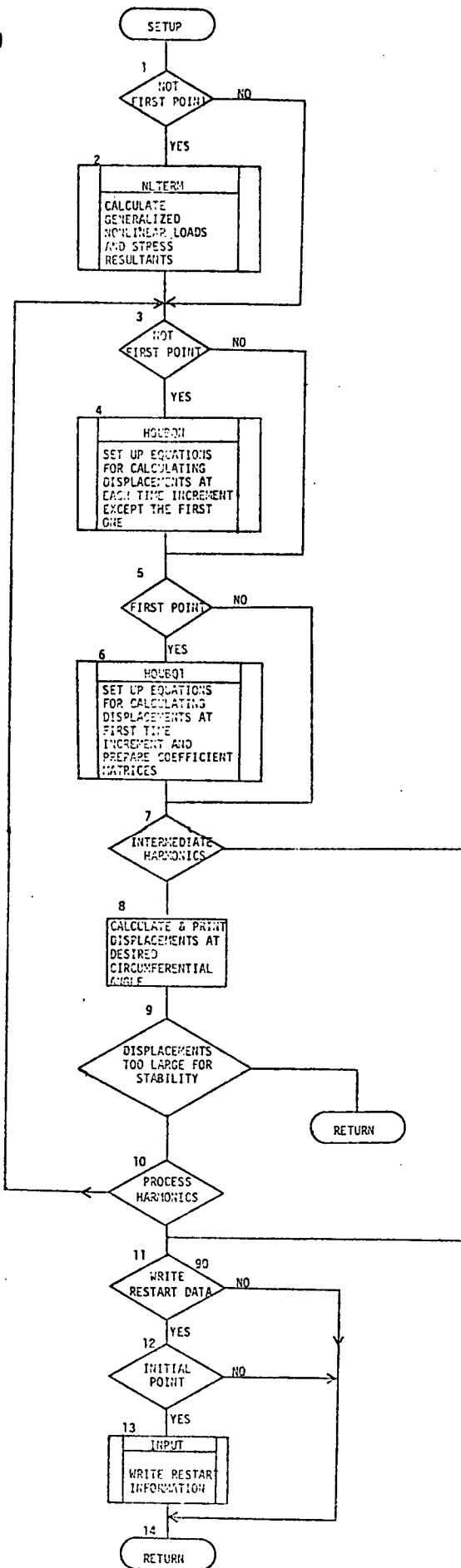


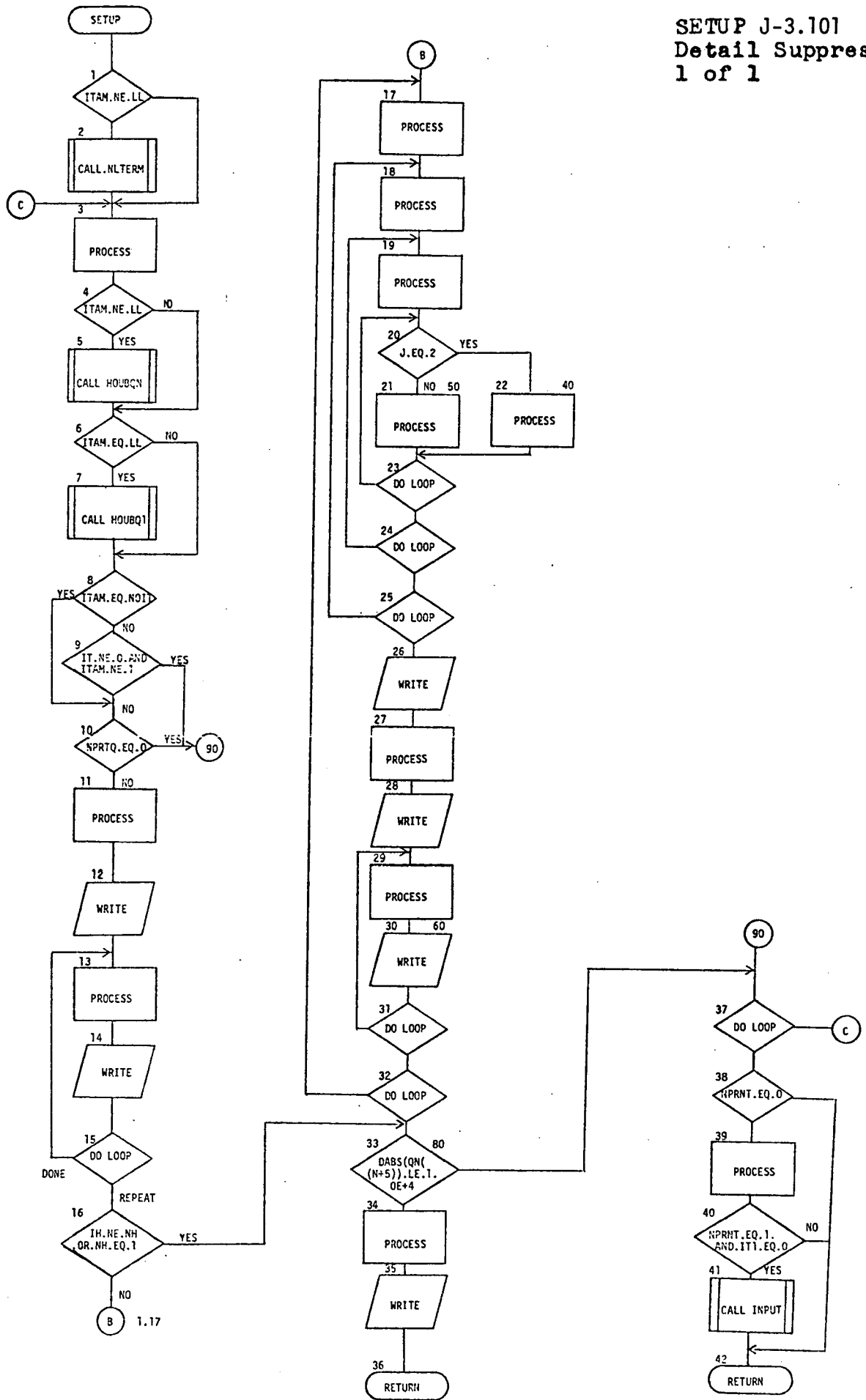
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0043          DO 50 J=1,4          DYN06160
0044             IF (J.EQ.2) GO TO 40 DYNC6170
0045             QLOAD(K+J)=QLOAD(K+J)+QN(NPK+J)*CS DYN06180
0046             GO TO 50          DYN06190
0047             QLOAD(K+J)=QLOAD(K+J)+QN(NPK+J)*SN DYN06200
0048          40          CONTINUE DYN06210
          C          DYN06213
0049             WRITE (6,110) ITAM,TPRNT DYN06220
0050             THETA1=THETA(IT)*180./3.14159 DYN06230
0051             WRITE (6,140) THETA1 DYN06240
          C          DYN06248
0052             DO 60 I=1,NNODES DYN06250
0053                K=4*(I-1) DYN06260
0054                WRITE (6,130) I,(QLOAD(K+J),J=1,4) DYN06270
0055          60          CONTINUE DYNC6272
          C          DYN06275
0056          70          CONTINUE DYN06280
          C          DYNC6283
          C1         DISPLACEMENTS TOO LARGE FOR STABILITY//NO(90) DYN06285
0057          80          IF (DABS(QN(N+5)).LE.1.0E+4) GO TO 90 DYN06330
0058             LARGE=1 DYN06340
0059             WRITE (6,150) ITAM,TIME DYN06350
0060             RETURN DYN06360
0061          90          CONTINUE DYNC6370
          C          DYN06373
          C1         WRITE RESTART DATA//NO(100) DYN06375
0062             IF (NPRNT.EQ.0) GO TO 100 DYN06380
0063             IT1=ITAM-(ITAM/NPRNT)*NPRNT DYN06390
0064             IF (NPRNT.EQ.1.AND.IT1.EQ.0) CALL INPUT (3) DYN06400
0065          100         RETURN DYN06410
          C          DYN06420
0066          110         FORMAT (1H1,30X,6HITAM =,I5,5X,6HTIME =,F12.4,13H MICROSECONDS//) DYN06430
0067          120         FORMAT (36X,22HDISPLACEMENTS OF NODES/38X,9HHARMONIC ,I5// DYNC6440
          1           6X,8HNODE NO.,6X,5HAXIAL,13X,10HTANGENTIAL,11X,6HRADIAL, DYN06450
          1           13X,7HANGULAR//) DYN06452
0068          130         FORMAT (I10,4D2C.8) DYNC6460
0069          140         FORMAT (25X,34HDISPLACEMENTS OF NODES AT THETA = ,F8.3, DYN06470
          1           9H DEGREES/38X,13HALL HARMONICS/ DYNC6480
          1           2X,8HNODE NO.,9X,5HAXIAL,12X,10HTANGENTIAL,12X,6HRADIAL, DYN06490
          1           13X,7HANGULAR//) DYN06492
0070          150         FORMAT (1H1,5X,4HITAM,I5,5X,4HTIME,E12.5// DYN06500
          1           6X,22HEXECUTION TERMINATED -, DYN06510
          1           33H DISPLACEMENTS GREATER THAN 1.E+4) DYN06512
0071          END          DYN06520

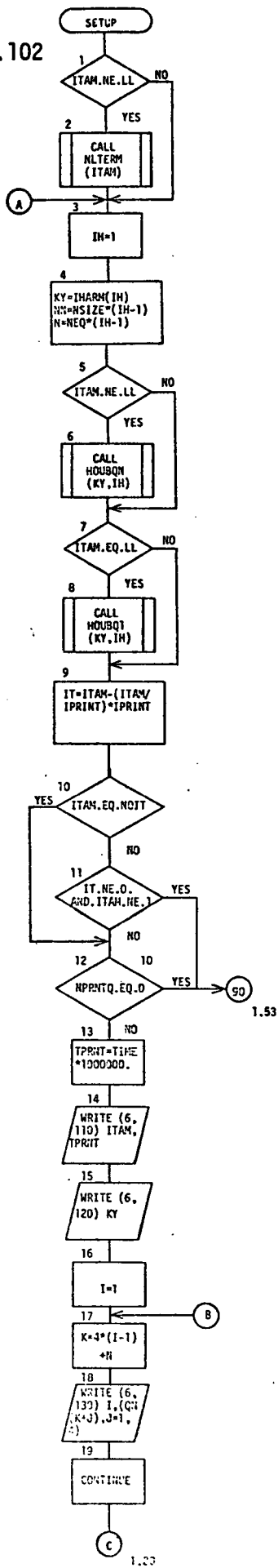
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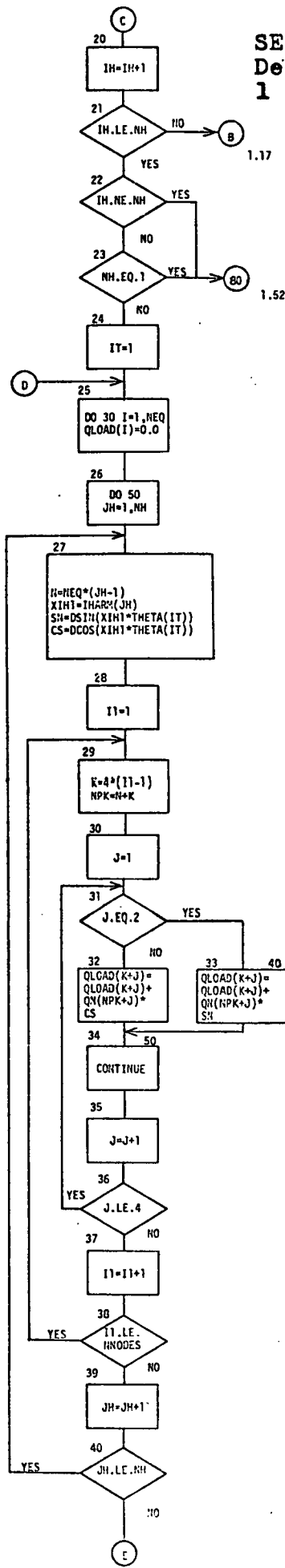


J-3.102

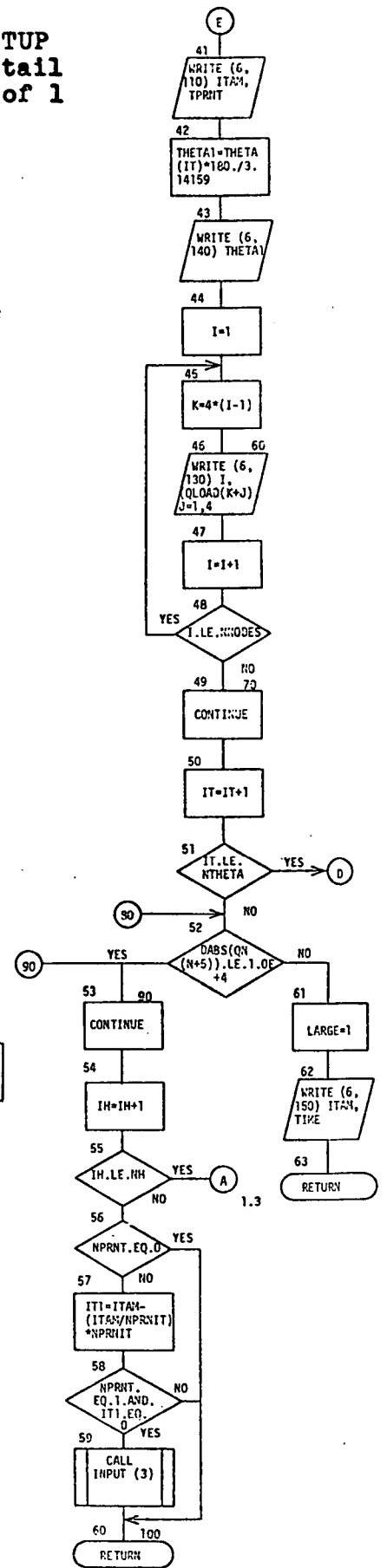


1.33

### SETUP Detail 1 of 1



1.41



1.3

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CE(SOLVEQ) DYN11242
C DYN11244
C DESCRIPTION - TO SOLVE THE SET OF SIMULTANEOUS LINEAR DYN11246
C EQUATIONS USING A MODIFIED GAUSS ELIMINATION TECHNIQUE DYN11248
C WHICH TAKES INTO ACCOUNT DIAGONAL SYMMETRY OF THE DYN11250
C SYSTEM. THIS ROUTINE WILL ACCEPT ONLY SPECIAL BANDED DYN11252
C MATRICES (I.E., FORMED BY OVERLAPPING SUBMATRICES). DYN11254
C ELEMENTS OF THE UPPER BAND ARE STORED IN A 1-DIMENSIONAL DYN11256
C ARRAY BY COLUMNS. ALL ZEROS THAT FALL OUTSIDE OF DYN11258
C THE SUB-MATRICES ARE SUPPRESSED. DYN11260
C DYN11262
C INPUT ARGUMENTS. DYN11264
C IH = HARMONIC KEY. DYN11266
C DYN11268
C OUTPUT ARGUMENTS. DYN11270
C SPR = ALIAS FOR QLOAD. DYN11272
C DYN11274
C EXTERNALS. DYN11276
C CALLED BY DYN11278
C HOUBQ1 DYN11280
C HOUBQN DYN11282
C DYN11284
0001 SUBROUTINE SOLVEQ (IH) DYN11286
0002 IMPLICIT REAL*8 (A-H,O-Z) DYN11288
0003 COMMON /SLVEEQ/ SPA(6550),SPR(204) DYN11290
0004 COMMON /CONST/ NH,NELEMS,NNODES,NSIZE,NPRNTQ,NEQ,NEQT,N,NN,NHNS, DYN11292
1 DT2,NPRNTL,NPRNTF,IDELF,IDCOE DYN11294
0005 COMMON /PRINT/ IPRINT,NOIT,LL DYN11296
0006 COMMON /CYCLE/ ITAM DYN11300
0007 DIMENSION A(1310), R(204) DYN11310
C1 GAUSSIAN ELIMINATION PASS DYN11312
C DYN11388
0008 DO 10 I=1,NSIZE DYN11390
0009 NNPI=NN+I DYN11400
0010 A(I)=SPA(NNPI) DYN11410
0011 10 CONTINUE DYN11412
C DYN11415
C DYN11418
0012 DO 20 I=1,NEQ DYN11420
0013 R(I)=SPR(I) DYN11430
0014 20 CONTINUE DYN11432
C DYN11435
0015 IF (ITAM.GT.(LL+1)) GO TO 100 DYN11440
0016 KEY=2 DYN11450
0017 A(2)=A(2)/A(1) DYN11460
0018 A(3)=A(3)-A(2)*A(2)*A(1) DYN11470
0019 A(5)=A(5)-A(4)*A(2) DYN11480
0020 A(8)=A(8)-A(7)*A(2) DYN11490

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0021		A(4)=A(4)/A(1)	DYN11500
0022		A(5)=A(5)/A(3)	DYN11510
0023		A(6)=A(6)-A(4)*A(4)*A(1)-A(5)*A(5)*A(3)	DYN11520
0024		A(9)=A(9)-A(7)*A(4)-A(8)*A(5)	DYN11530
0025		A(7)=A(7)/A(1)	DYN11540
0026		A(8)=A(8)/A(3)	DYN11550
0027		A(9)=A(9)/A(6)	DYN11560
0028		A(10)=A(10)-A(7)*A(7)*A(1)-A(8)*A(8)*A(3)-A(9)*A(9)*A(6)	DYN11570
0029		R(2)=R(2)-R(1)*A(2)	DYN11580
0030		R(3)=R(3)-R(1)*A(4)-R(2)*A(5)	DYN11590
0031		R(4)=R(4)-R(1)*A(7)-R(2)*A(8)-R(3)*A(9)	DYN11600
	C		DYN11608
0032		DO 90 K=1,NELEMS	DYN11610
0033		I=1+(K-1)*26	DYN11620
0034		J=5+(K-1)*4	DYN11630
0035		GO TO (40,30), KEY	DYN11640
0036	30	AM10=A(I-10)	DYN11650
0037		AM9=A(I-9)	DYN11660
0038		AM8=A(I-8)	DYN11670
0039		AM7=A(I-7)	DYN11680
0040		AM6=A(I-6)	DYN11690
0041		AM5=A(I-5)	DYN11700
0042		GO TO 50	DYN11710
0043	40	AM10=A(I-22)	DYN11720
0044		AM9=A(I-17)	DYN11730
0045		AM8=A(I-16)	DYN11740
0046		AM7=A(I-11)	DYN11750
0047		AM6=A(I-10)	DYN11760
0048		AM5=A(I-9)	DYN11770
0049	50	AM4=A(I-4)	DYN11780
0050		AM3=A(I-3)	DYN11790
0051		AM2=A(I-2)	DYN11800
0052		AM1=A(I-1)	DYN11810
0053		A0=A(I)	DYN11820
0054		A1=A(I+1)	DYN11830
0055		A2=A(I+2)	DYN11840
0056		A3=A(I+3)	DYN11850
0057		A4=A(I+4)	DYN11860
0058		A5=A(I+5)	DYN11870
0059		A6=A(I+6)	DYN11880
0060		A7=A(I+7)	DYN11890
0061		A8=A(I+8)	DYN11900
0062		A9=A(I+9)	DYN11910
0063		A10=A(I+10)	DYN11920
0064		A11=A(I+11)	DYN11930
0065		A12=A(I+12)	DYN11940
0066		A13=A(I+13)	DYN11950
0067		A14=A(I+14)	DYN11960

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0068      A15=A(I+15)          DYN11970
0069      A16=A(I+16)          DYN11980
0070      A17=A(I+17)          DYN11990
0071      A18=A(I+18)          DYN12000
0072      A19=A(I+19)          DYN12010
0073      A20=A(I+20)          DYN12020
0074      A21=A(I+21)          DYN12030
0075      A22=A(I+22)          DYN12040
0076      A23=A(I+23)          DYN12050
0077      A24=A(I+24)          DYN12060
0078      A25=A(I+25)          DYN12070
0079      A1=A1-A0*AM9         DYN12080
0080      A6=A6-A5*AM9         DYN12090
0081      A12=A12-A11*AM9      DYN12100
0082      A19=A19-A18*AM9      DYN12110
0083      A2=A2-A0*AM7-A1*AM6  DYN12120
0084      A7=A7-A5*AM7-A6*AM6  DYN12130
0085      A13=A13-A11*AM7-A12*AM6 DYN12140
0086      A20=A20-A18*AM7-A19*AM6 DYN12150
0087      A3=A3-A0*AM4-A1*AM3-A2*AM2 DYN12160
0088      A8=A8-A5*AM4-A6*AM3-A7*AM2 DYN12170
0089      A14=A14-A11*AM4-A12*AM3-A13*AM2 DYN12180
0090      A21=A21-A18*AM4-A19*AM3-A20*AM2 DYN12190
0091      AC=A0/AM10          DYN12200
0092      A1=A1/AM8           DYN12210
0093      A2=A2/AM5           DYN12220
0094      A4=A4-A0*A0*AM10-A1*A1*AM8-A2*A2*AM5-A3*A3*AM1 DYN12230
0095      A9=A9-A5*A0-A6*A1-A7*A2-A8*A3 DYN12240
0096      A15=A15-A11*A0-A12*A1-A13*A2-A14*A3 DYN12250
0097      A22=A22-A18*A0-A19*A1-A20*A2-A21*A3 DYN12260
0098      R(J)=R(J)-R(J-4)*A0-R(J-3)*A1-R(J-2)*A2-R(J-1)*A3 DYN12270
0099      A5=A5/AM10          DYN12280
0100      A6=A6/AM8           DYN12290
0101      A7=A7/AM5           DYN12300
0102      A8=A8/AM1           DYN12310
0103      A9=A9/A4            DYN12320
0104      A10=A10-A5*A5*AM10-A6*A6*AM8-A7*A7*AM5-A8*A8*AM1-A9*A9*A4 DYN12330
0105      A16=A16-A5*A11-A6*A12-A7*A13-A8*A14-A9*A15 DYN12340
0106      A23=A23-A5*A18-A6*A19-A7*A20-A8*A21-A9*A22 DYN12350
0107      R(J+1)=R(J+1)-R(J-4)*A5-R(J-3)*A6-R(J-2)*A7-R(J-1)*A8-R(J)*A9 DYN12360
0108      A11=A11/AM10        DYN12370
0109      A12=A12/AM8         DYN12380
0110      A13=A13/AM5         DYN12390
0111      A14=A14/AM1         DYN12400
0112      A15=A15/A4          DYN12410
0113      A16=A16/A10         DYN12420
0114      A17=A17-A11*A11*AM10-A12*A12*AM8-A13*A13*AM5-A14*A14*AM1-A15* DYN12430
          A15*A4-A16*A16*A10 DYN12440
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0115      A24=A24-A11*A18-A12*A19-A13*A20-A14*A21-A15*A22-A16*A23      DYN12450
0116      R(J+2)=R(J+2)-R(J-4)*A11-R(J-3)*A12-R(J-2)*A13-R(J-1)*A14-R(J)*A15-R(J+1)*A16      DYN12460
          1      A18=A18/AM10      DYN12470
0117      A18=A18/AM10      DYN12480
0118      A19=A19/AM8      DYN12490
0119      A20=A20/AM5      DYN12500
0120      A21=A21/AM1      DYN12510
0121      A22=A22/A4      DYN12520
0122      A23=A23/A10      DYN12530
0123      A24=A24/A17      DYN12540
0124      A25=A25-A18*A18*AM10-A19*A19*AM8-A20*A20*AM5-A21*A21*AM1-A22*      DYN12550
          1      A22*A4-A23*A23*A10-A24*A24*A17      DYN12560
0125      R(J+3)=R(J+3)-R(J-4)*A18-R(J-3)*A19-R(J-2)*A20-R(J-1)*A21-R(J)*A22-R(J+1)*A23-R(J+2)*A24      DYN12570
          1      GO TO (70,60), KEY      DYN12580
0126      GO TO (70,60), KEY      DYN12590
0127      60      A(I-10)=AM10      DYN12600
0128      A(I-9)=AM9      DYN12610
0129      A(I-8)=AM8      DYN12620
0130      A(I-7)=AM7      DYN12630
0131      A(I-6)=AM6      DYN12640
0132      A(I-5)=AM5      DYN12650
0133      KEY=1      DYN12660
0134      GO TO 80      DYN12670
0135      70      A(I-22)=AM10      DYN12680
0136      A(I-17)=AM9      DYN12690
0137      A(I-16)=AM8      DYN12700
0138      A(I-11)=AM7      DYN12710
0139      A(I-10)=AM6      DYN12720
0140      A(I-9)=AM5      DYN12730
0141      80      A(I-4)=AM4      DYN12740
0142      A(I-3)=AM3      DYN12750
0143      A(I-2)=AM2      DYN12760
0144      A(I-1)=AM1      DYN12770
0145      A(I)=A0      DYN12780
0146      A(I+1)=A1      DYN12790
0147      A(I+2)=A2      DYN12800
0148      A(I+3)=A3      DYN12810
0149      A(I+4)=A4      DYN12820
0150      A(I+5)=A5      DYN12830
0151      A(I+6)=A6      DYN12840
0152      A(I+7)=A7      DYN12850
0153      A(I+8)=A8      DYN12860
0154      A(I+9)=A9      DYN12870
0155      A(I+10)=A10      DYN12880
0156      A(I+11)=A11      DYN12890
0157      A(I+12)=A12      DYN12900
0158      A(I+13)=A13      DYN12910
0159      A(I+14)=A14      DYN12920
    
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U

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0160          A(I+15)=A15                      DYN12930
0161          A(I+16)=A16                      DYN12940
0162          A(I+17)=A17                      DYN12950
0163          A(I+18)=A18                      DYN12960
0164          A(I+19)=A19                      DYN12970
0165          A(I+20)=A20                      DYN12980
0166          A(I+21)=A21                      DYN12990
0167          A(I+22)=A22                      DYN13000
0168          A(I+23)=A23                      DYN13010
0169          A(I+24)=A24                      DYN13020
0170          A(I+25)=A25                      DYN13030
0171          90 CONTINUE                      DYN13032
C
0172          GO TO 120                        DYN13040
0173          100 CONTINUE                     DYN13050
0174          R(2)=R(2)-R(1)*A(2)             DYN13060
0175          R(3)=R(3)-R(1)*A(4)-R(2)*A(5)   DYN13070
0176          R(4)=R(4)-R(1)*A(7)-R(2)*A(8)-R(3)*A(9) DYN13080
C
0177          DO 110 K=1,NELEMS                DYN13090
0178             N=4*(K+1)-3                   DYN13100
0179             L=26*K-15                     DYN13110
0180             R(N)=R(N)-R(N-4)*A(L)-R(N-3)*A(L+1)-R(N-2)*A(L+2)-R(N-1)*A(L+3) DYN13120
0181             R(N+1)=R(N+1)-R(N-4)*A(L+5)-R(N-3)*A(L+6)-R(N-2)*A(L+7)-
1              R(N-1)*A(L+8)-R(N)*A(L+9)      DYN13132
0182             R(N+2)=R(N+2)-R(N-4)*A(L+11)-R(N-3)*A(L+12)-R(N-2)*A(L+13)-
1              R(N-1)*A(L+14)-R(N)*A(L+15)-R(N+1)*A(L+16) DYN13136
0183             R(N+3)=R(N+3)-R(N-4)*A(L+18)-R(N-3)*A(L+19)-R(N-2)*A(L+20)-
1              R(N-1)*A(L+21)-R(N)*A(L+22)-R(N+1)*A(L+23)-R(N+2)*
1              A(L+24)                        DYN13142
0184          110 CONTINUE                     DYN13190
C
0185          120 CONTINUE                     DYN13193
C1          BACK SUBSTITUTIONS                DYN13200
0186             N=26*NELEMS+10                DYN13202
0187             M=(NELEMS+1)*4                DYN13220
0188             R(M)=R(M)/A(N)                DYN13230
0189             R(M-1)=R(M-1)/A(N-8)-R(M)*A(N-1) DYN13240
0190             R(M-2)=R(M-2)/A(N-15)-R(M-1)*A(N-9)-R(M)*A(N-2) DYN13250
0191             R(M-3)=R(M-3)/A(N-21)-R(M-2)*A(N-16)-R(M-1)*A(N-10)-R(M)*A(N-3) DYN13260
C
0192          DO 150 K=1,NELEMS                DYN13270
0193             L=26*(NELEMS+1-K)-15           DYN13278
0194             N=4*(NELEMS+1-K)-3            DYN13280
0195             IF (L-11) 130,140,130         DYN13280
0196             130 R(N+3)=R(N+3)/A(L-1)-R(N+4)*A(L+3)-R(N+5)*A(L+8)-R(N+6)*
1              A(L+14)-R(N+7)*A(L+21)        DYN13290
0197             R(N+2)=R(N+2)/A(L-9)-R(N+3)*A(L-2)-R(N+4)*A(L+2)-R(N+5)*
1              A(L+14)-R(N+7)*A(L+21)        DYN13300
1              A(L+14)-R(N+7)*A(L+21)        DYN13310
1              A(L+14)-R(N+7)*A(L+21)        DYN13320
1              A(L+14)-R(N+7)*A(L+21)        DYN13330
1              A(L+14)-R(N+7)*A(L+21)        DYN13340

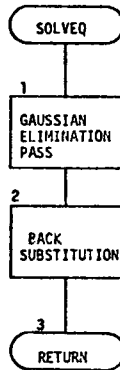
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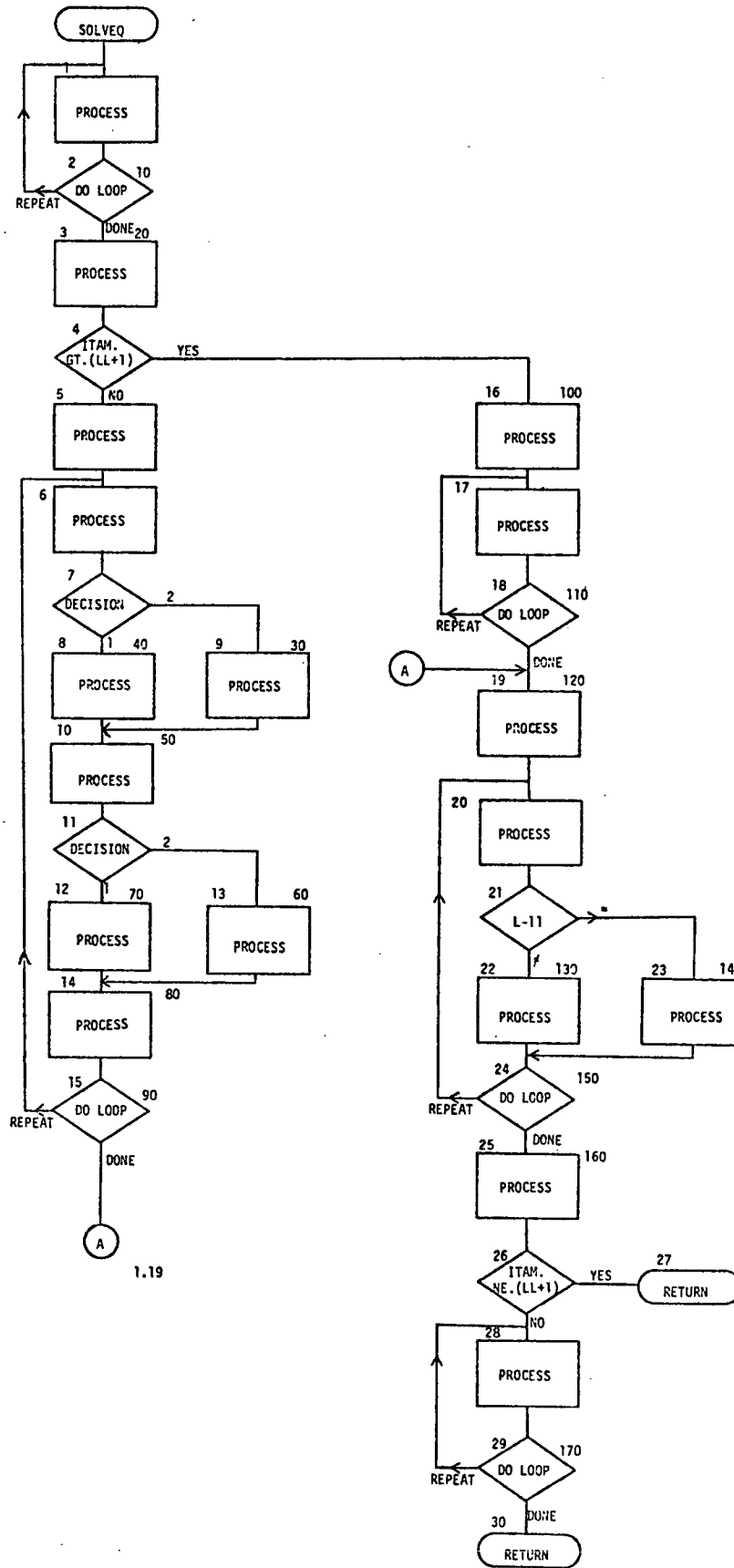


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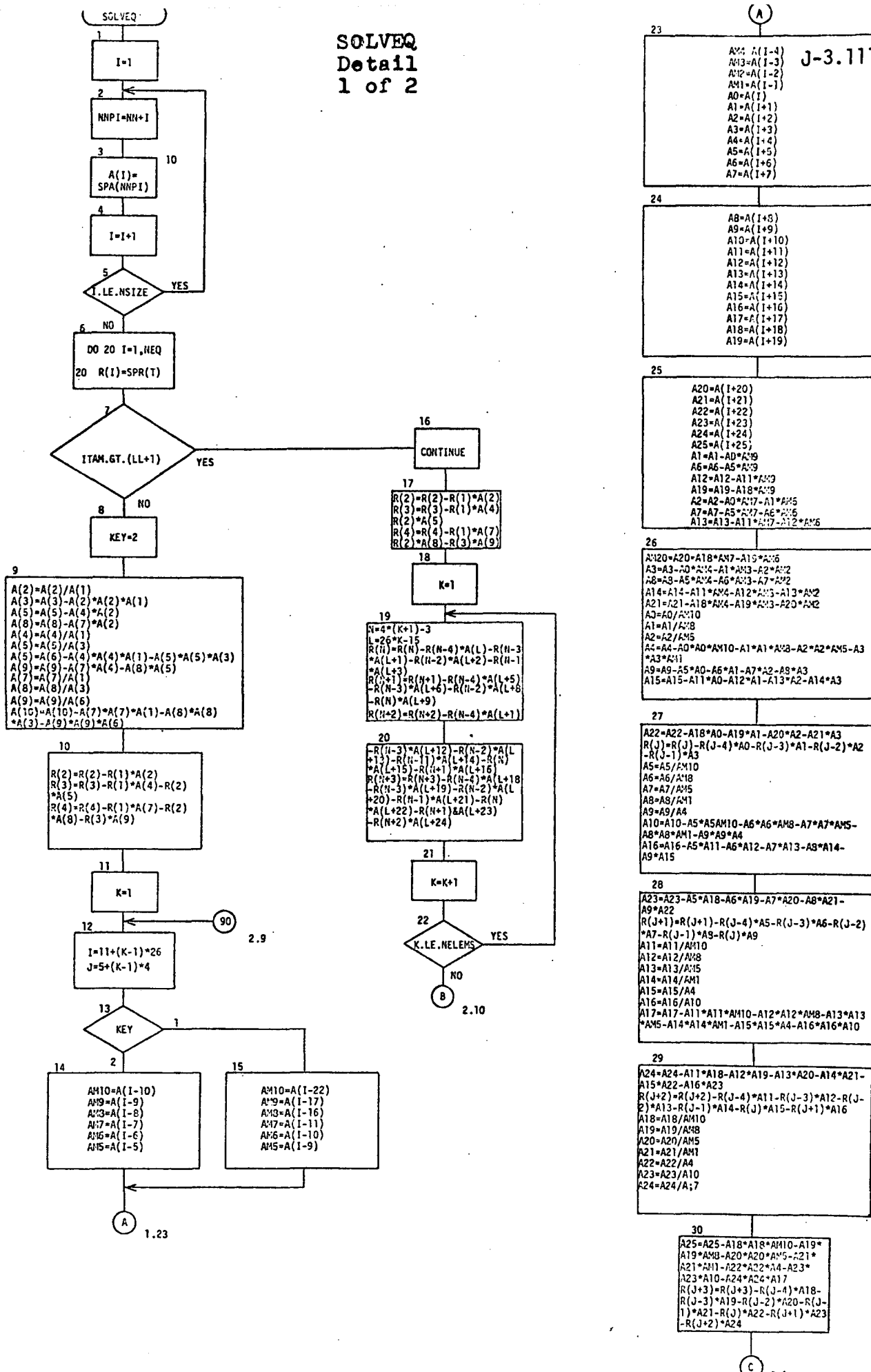
1          A(L+7)-R(N+6)*A(L+13)-R(N+7)*A(L+20)          DYN13350
0198      R(N+1)=R(N+1)/A(L-16)-R(N+2)*A(L-10)-R(N+3)*A(L-3)-R(N+4)*  DYN13360
1          A(L+1)-R(N+5)*A(L+6)-R(N+6)*A(L+12)-R(N+7)*A(L+19)  DYN13370
0199      R(N)=R(N)/A(L-22)-R(N+1)*A(L-17)-R(N+2)*A(L-11)-R(N+3)*A(L-4)-  DYN13380
1          R(N+4)*A(L)-R(N+5)*A(L+5)-R(N+6)*A(L+11)-R(N+7)*  DYN13390
1          A(L+18)          DYN13391
0200      GO TO 150          DYN13400
0201      140 R(N+3)=R(N+3)/A(L-1)-R(N+4)*A(L+3)-R(N+5)*A(L+8)-R(N+6)*  DYN13410
1          A(L+14)-R(N+7)*A(L+21)          DYN13420
0202      R(N+2)=R(N+2)/A(L-5)-R(N+3)*A(L-2)-R(N+4)*A(L+2)-R(N+5)*A(L+7)-  DYN13430
1          R(N+6)*A(L+13)-R(N+7)*A(L+20)  DYN13440
0203      R(N+1)=R(N+1)/A(L-8)-R(N+2)*A(L-6)-R(N+3)*A(L-3)-R(N+4)*A(L+1)-  DYN13450
1          R(N+5)*A(L+6)-R(N+6)*A(L+12)-R(N+7)*A(L+19)  DYN13460
0204      R(N)=R(N)/A(L-10)-R(N+1)*A(L-9)-R(N+2)*A(L-7)-R(N+3)*A(L-4)-  DYN13470
1          R(N+4)*A(L)-R(N+5)*A(L+5)-R(N+6)*A(L+11)-R(N+7)*  DYN13480
1          A(L+18)          DYN13481
0205      150 CONTINUE          DYN13490
C          DYN13493
C          DYN13498
0206      DO 160 I=1,NEQ          DYN13500
0207          SPR(I)=R(I)          DYN13510
0208      160 CONTINUE          DYN13512
C          DYN13515
0209      IF (ITAM.NE.(LL+1)) RETURN          DYN13520
C          DYN13528
0210      DO 170 I=1,NSIZE          DYN13530
0211          NNPI=NN+I          DYN13540
0212          SPA(NNPI)=A(I)          DYN13550
0213      170 CONTINUE          DYN13552
C          DYN13555
0214      RETURN          DYN13560
0215      END          DYN13570

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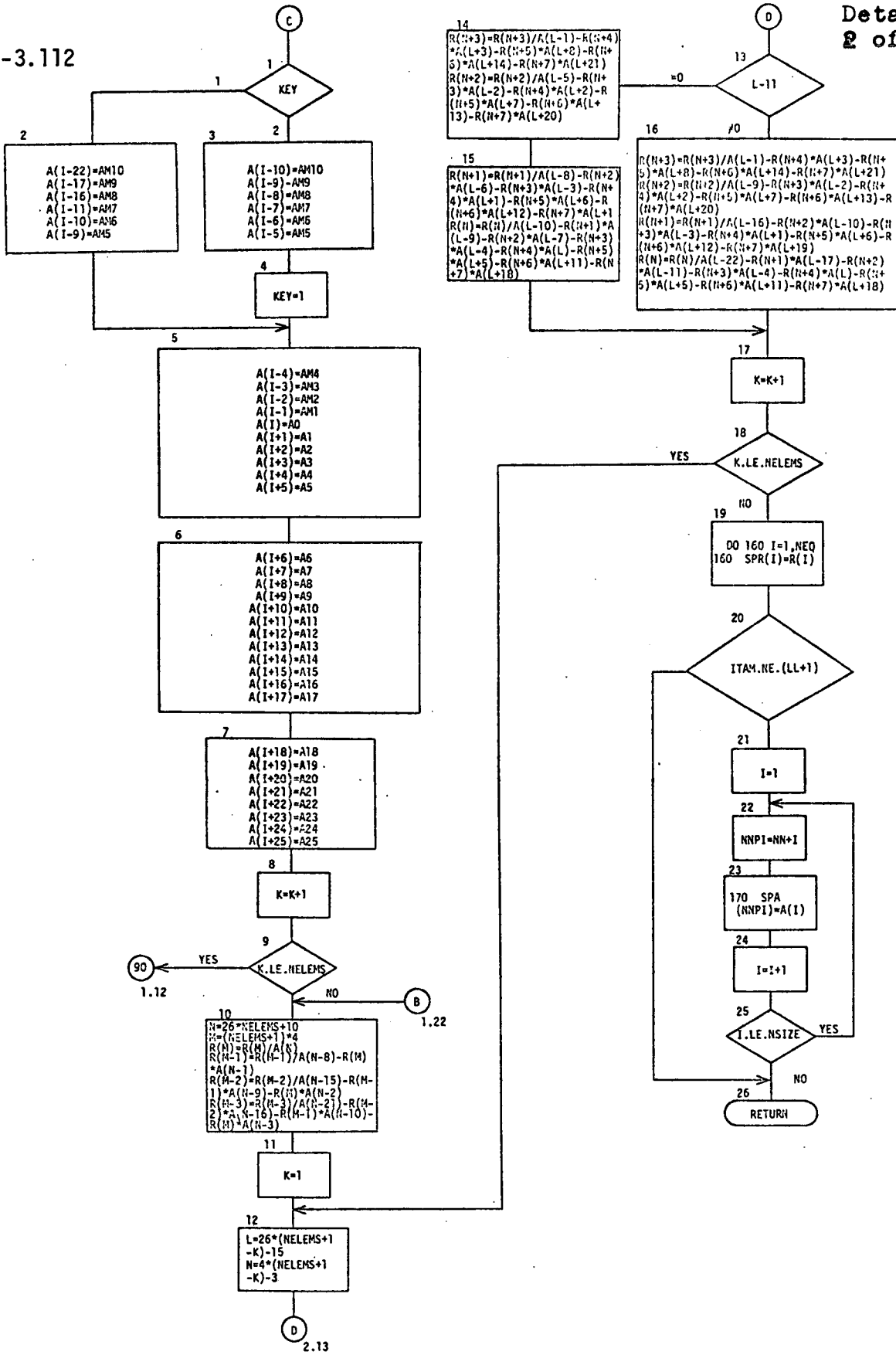




SOLVEQ  
Detail  
1 of 2



J-3.112



90 1.12

B 1.22

D 2.13

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CE(STRESS)
C
C DESCRIPTION - TO CALCULATE THE MIDSURFACE STRESSES AND
C STRESS RESULTANTS ON THE UPPER AND LOWER SHELL SURFACES.
C
C INPUT ARGUMENTS.
C DTH = MATRIX OF FOURIER COEFFICIENTS FOR THE CIRCUMFERENTIAL
C TEMPERATURE GRADIENT DISTRIBUTION.
C IHARM = MATRIX OF HARMONIC NUMBERS FOR WHICH DISPLACEMENTS
C AND/OR STRESSES WILL BE CALCULATED.
C ITAM = NUMBER OF TIME CYCLE.
C I1 = PRINT FLAG FOR STRESSES.
C NH = TOTAL NUMBER OF HARMONICS USED IN THE DYNAMIC ANALYSIS.
C NTHETA = THE NUMBER OF CIRCUMFERENTIAL ANGLES AT WHICH DISPLACE-
C MENTS AND/OR STRESSES ARE TO BE CALCULATED.
C TH = MATRIX WHOSE ELEMENTS ARE THE FOURIER COEFFICIENTS
C OF THE CIRCUMFERENTIAL TEMPERATURE DISTRIBUTION.
C THETA = MATRIX CONTAINING CIRCUMFERENTIAL ANGLES AT WHICH
C STRESSES AND/OR DISPLACEMENTS ARE TO BE CALCULATED.
C
C EXTERNALS.
C CALLED BY
C NLTERM
C
0001 SUBROUTINE STRESS (I1,ITAM)
0002 IMPLICIT REAL*8 (A-H,O-Z)
0003 COMMON /EES/ ES(5),ET(5),EST(5),E13(5),E23(5)
0004 COMMON /CONST/ NH,NELEMS,NNODES,NSIZE,NPRNTQ,NEQ,NEQT,N,NN,NHNS,
1 DT2,NPRNTL,NPRNTF,IDEFL,IDCDE
0005 COMMON /THETAS/ THETA(20),NTHETA,NCLCST,NSTRSS
0006 COMMON /GEOM/ FNU1(50),FNU2(50),E1(50),E2(50),G(50),T(50),
1 SINE(51),COSINE(51),SINM(50),COSM(50),R(50),PH(50),
1 PHP(50),ARCL(50)
0007 COMMON /GCD/ CC1,CC2,DD1,DD2,GG1,GG2
0008 COMMON /HARM/ NHP,IHARM(5)
0009 COMMON /TMFT/ TOTIME,DELTE,TIME,TO,T1
0010 COMMON /QS/ QN(1020),QN1(1020),FORCE(2040),QP(1020),QP1(1020),
1 QN2(1020)
0011 COMMON /THER/ TH(50,5,2),DTH(50,5,2),ALS(50),ALT(50)
0012 DIMENSION BSTRMS(20),BSTRMT(20),BSTMST(20),BTTMST(20),
1 BSTTMT(20)
0013 ITAM1=ITAM-1
0014 TM1=TIME-DELTE
0015 TPRINT=TM1*1000000.
0016 IF (I1.EQ.1) WRITE (6,50) ITAM1,TPRINT
C100 PROCESS ANGLES
C
0017 DO 40 I=1,NTHETA

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DYN13582  
DYN13584  
DYN13586  
DYN13588  
DYN13590  
DYN13592  
DYN13594  
DYN13596  
DYN13598  
DYN13600  
DYN13602  
DYN13604  
DYN13606  
DYN13608  
DYN13610  
DYN13612  
DYN13614  
DYN13616  
DYN13618  
DYN13620  
DYN13622  
DYN13624  
DYN13626  
DYN13628  
DYN13630  
DYN13632  
DYN13634  
DYN13636  
DYN13638  
DYN13640  
DYN13642  
DYN13650  
DYN13652  
DYN13660  
DYN13670  
DYN13680  
DYN13690  
DYN13700  
DYN13710  
DYN13720  
DYN13730  
DYN13780  
DYN13790  
DYN13800  
DYN13810  
DYN13812  
DYN13818  
DYN13820

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0018      ESU=0.0                      DYN13830
0019      ETU=0.0                      DYN13840
0020      ESTU=0.0                     DYN13850
0021      E13U=0.0                    DYN13860
0022      E23U=0.0                    DYN13870
0023      CHIS=0.0                     DYN13880
0024      CHIT=0.0                     DYN13890
0025      CHIST=0.0                    DYN13900
0026      CTHIS=0.0                   DYN13910
0027      CTHIT=0.0                   DYN13920
0028      CTHIST=0.0                  DYN13930
0029      ESUT=0.0                     DYN13940
          C1 DO      PROCESS HARMONICS DYN13950
          C          DYN13958
0030      DO 10 IH=1,NH                DYN13960
0031      XIH1=IHARM(IH)               DYN13970
0032      CS=DCOS(XIH1*THETA(I))        DYN13980
0033      SN=DSIN(XIH1*THETA(I))        DYN13990
0034      K=4*(I1-1)+NEQ*(IH-1)         DYN14000
0035      THT=TH(I1,IH,1)+(TH(I1,IH,2)-TH(I1,IH,1))*(TIME-T0)/(T1-T0) DYN14010
0036      DTHT=DTH(I1,IH,1)+(DTH(I1,IH,2)-DTH(I1,IH,1))*(TIME-T0)/(T1- DYN14020
          T0)                            DYN14021
          C1 1      CALCULATION OF LINEAR STRAINS AND ROTATIONS DYN14023
0037      ESU=ESU+ES(IH)*CS            DYN14040
0038      ETU=ETU+ET(IH)*CS            DYN14050
0039      ESTU=ESTU+EST(IH)*SN          DYN14060
0040      E13U=E13U+E13(IH)*CS         DYN14070
0041      E23U=E23U+E23(IH)*SN         DYN14080
0042      ESUT=ESUT+ALS(I1)*THT*CS     DYN14090
0043      ETUT=ETUT+ALT(I1)*THT*CS     DYN14100
          C1      CALCULATION OF CHANGES IN CURVATURE DYN14102
0044      QB3=-QN(K+1)*SINE(I1)+QN(K+3)*COSINE(I1) DYN14120
0045      QB7=-QN(K+5)*SINE(I1+1)+QN(K+7)*COSINE(I1+1) DYN14130
0046      CHIS1=(QN(K+4)-QN(K+8))/ARCL(I1) DYN14140
0047      CHIS2=ALS(I1)*DTHT            DYN14150
0048      CHIS=CHIS+(CHIS1-CHIS2)*CS   DYN14160
0049      CHIT1=(-XIH1*E23(IH)-SINM(I1)*E13(IH))/R(I1) DYN14170
0050      CHIT2=ALT(I1)*DTHT            DYN14180
0051      CHIT=CHIT+(CHIT1-CHIT2)*CS   DYN14190
0052      CHIST1=(XIH1*E13(IH)+SINM(I1)*E23(IH)-XIH1*SINM(I1)*(QB3+ DYN14200
          QB7)/(2.*R(I1))+XIH1*(QB7-QB3)/ARCL(I1)+(QN(K+6)- DYN14210
          QN(K+2))*COSM(I1)/ARCL(I1)-(QN(K+6)+QN(K+2))* DYN14220
          SINM(I1)*(COSM(I1)/(2.*R(I1))+PHP(I1)/2.))/R(I1) DYN14230
0053      CHIST=CHIST+CHIST1*SN         DYN14240
0054      CTHIS=CTHIS+XIH1*(CHIS1-CHIS2)*(-SN) DYN14250
0055      CTHIT=CTHIT+XIH1*(CHIT1-CHIT2)*(-SN) DYN14260
0056      CTHIST=CTHIST+XIH1*CHIST1*CS DYN14270
0057      10      CONTINUE              DYN14280

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C
C1      CALCULATION OF MIDSURFACE STRAINS
0058      EPS=ESU+.5*E13U**2-ESUT
0059      EPT=ETU+.5*E23U**2-ETUT
0060      EPST=ESTU+E13U*E23U
C1      CALCULATION OF STRESS AND MOMENT RESULTANTS
0061      STRNS=CC1*EPS+FNU1(I1)*CC1*EPT
0062      STRNT=FNU2(I1)*CC2*EPS+CC2*EPT
0063      STRNST=GG1*EPST
0064      STRMS=DD1*CHIS+FNU1(I1)*DD1*CHIT
0065      STRMT=FNU2(I1)*DD2*CHIS+DD2*CHIT
0066      STRMST=GG2*CHIST
C1      CALCULATE STRESSES ON THE INNER AND OUTER SURFACES
0067      C1ST=1.0/T(I1)
0068      C2ST=6.0/T(I1)**2
0069      BSU=C1ST*STRNS+C2ST*STRMS
0070      BTU=C1ST*STRNT+C2ST*STRMT
0071      BSTU=C2ST*STRMST+C1ST*STRNST
0072      BSL=C1ST*STRNS-C2ST*STRMS
0073      BTL=C1ST*STRNT-C2ST*STRMT
0074      BSTL=-C2ST*STRMST+C1ST*STRNST
C1      CALCULATE SHEAR RESULTANTS AND PRINT OUTPUT INFORMATION
0075      STTRMT=FNU2(I1)*DD2*CTHIS+DD2*CTHIT
0076      STTMST=GG2*CTHIST
0077      THETA1=THETA(I1)*180./3.14159
0078      IF (I1.NE.1) GO TO 20
0079      WRITE (6,60) I1,THETA1,STRNS,STRNT,STRNST,STRMS,STRMT,STRMST,
1          BSU,BTU,BSTU,BSL,BTL,BSTL
0080      GO TO 30
0081      20 RAV=(R(I1)+R(I1-1))/2.
0082      PAV=(PH(I1)+PH(I1-1))/2.
0083      AAV=(ARCL(I1)+ARCL(I1-1))/2.
0084      SHRS=1./RAV*((R(I1)*STRMS-R(I1-1)*BSTRMS(I1))/AAV+(STTMST+
1          BTTMST(I1))/2.-DSIN(PAV)*(STRMT+BSTRMT(I1))/2.)
0085      SHRT=1./RAV*((R(I1)*STRMST-R(I1-1)*BSTMST(I1))/AAV+(STTRMT+
1          BSTTMT(I1))/2.+DSIN(PAV)*(STRMST+BSTMST(I1))/2.)
0086      WRITE (6,70) I1,THETA1,STRNS,STRNT,STRNST,STRMS,STRMT,STRMST,
1          SHRS,SHRT,BSU,BTU,BSTU,BSL,BTL,BSTL
0087      30 BSTRMS(I)=STRMS
0088      BSTRMT(I)=STRMT
0089      BSTMST(I)=STRMST
0090      BTTMST(I)=STTMST
0091      BSTTMT(I)=STTRMT
0092      40 CONTINUE
C
0093      RETURN
C
0094      50 FORMAT (I1,3X,6HITAM =,I5,3X,6HTIME =,F12.4,13H MICROSECONDS,/)
    
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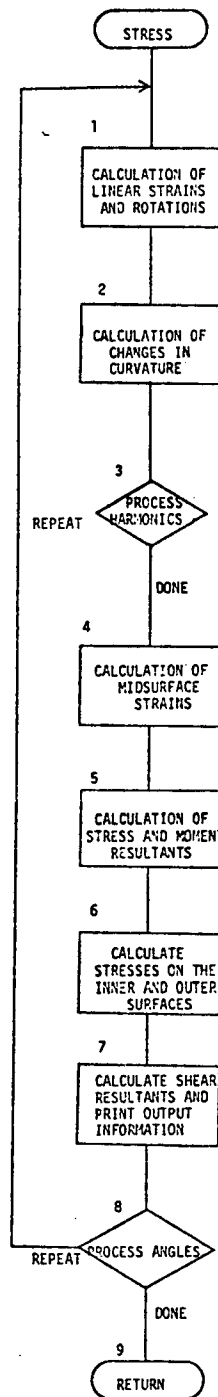
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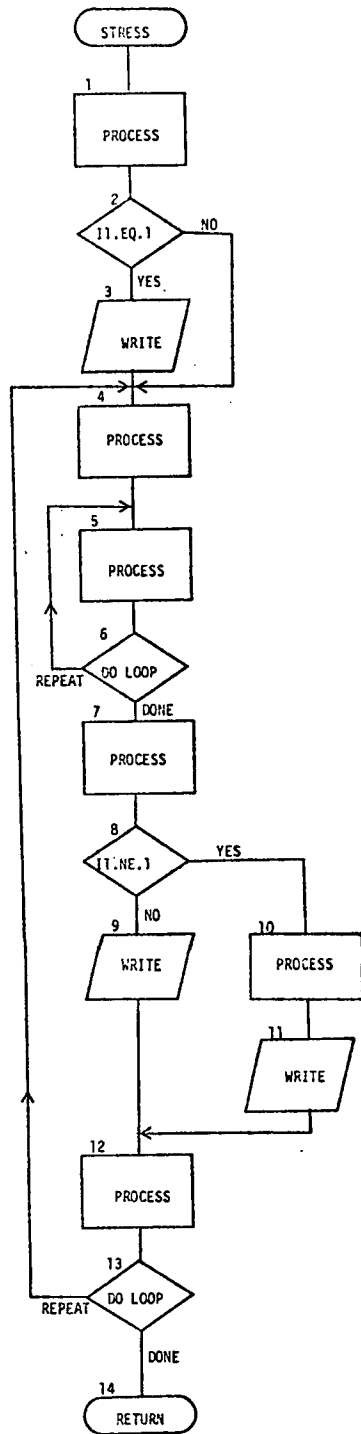


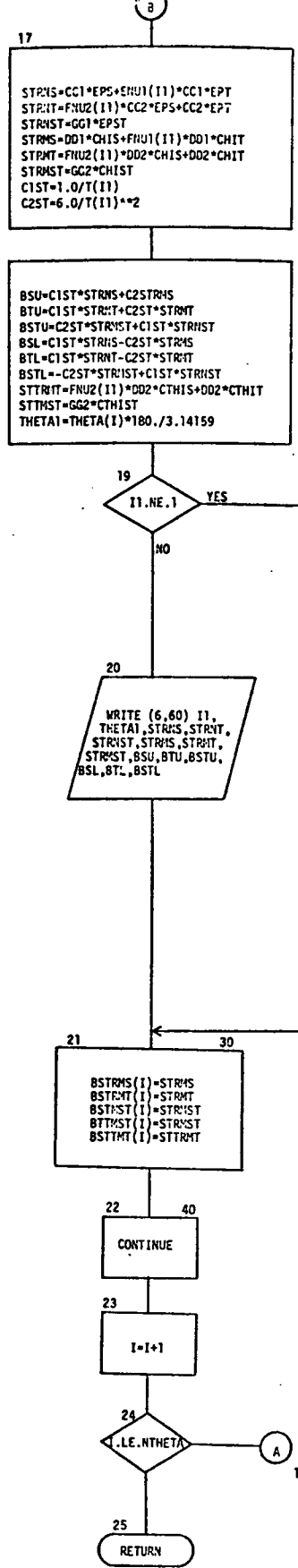
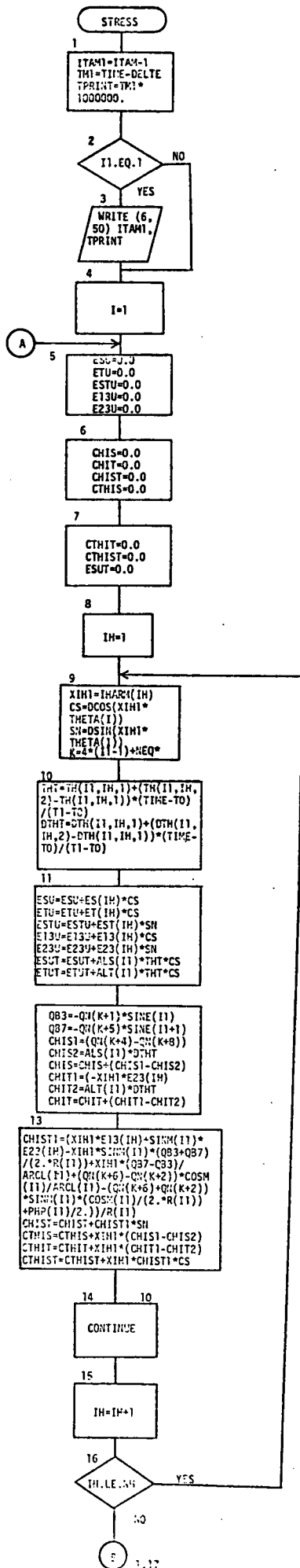
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1      47X,33HSTRESSES AND STRESS RESULTANTS,/ DYN14750
2      25X,17HFORCE RESULTANTS,31X,18HMOMFNT RESULTANTS,18X, DYN14760
2      17HSHEAR RESULTANTS/ DYN14770
2      19X,4HN(S),11X,4HN(T),11X,5HN(ST),10X,4HM(S),11X,4HM(T), DYN14780
2      11X,5HM(ST),10X,4HQ(S),11X,4HQ(T)//12H ELEM THETA,/ DYN14790
3      53H NO (DEG) ***** OUTER SURFACE STRESSES ,DYN14800
3      51H ***** INNER SURFACE STRESSES *****,/ DYN14810
4      15X,41H* SIGMA(S) SIGMA(T) SIGMA(ST), DYN14820
4      47H SIGMA(S) SIGMA(T) SIGMA(ST) */) DYN14822
0095 60 FORMAT (14,F8.2,6(1PD15.4),30H XXXX XXXX ,/ DYN14830
1      12H STRESSES **,6(1PD15.4)) DYN14840
0096 70 FORMAT (14,F8.2,8(1PD15.4),/,12H STRESSES **,6(1PD15.4)) DYN14850
0097 END DYN14860

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1.5

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C     CE(TFORCE)                                DYN16852
C     C                                          DYN16854
C     DESCRIPTION - TO CALCULATE THE LINEAR THERMAL LOADS ON THE SHELL STRUCTURE. DYN16856
C     C                                          DYN16858
C     C                                          DYN16860
C     INPUT ARGUMENTS.                          DYN16862
C     ARCL = MATRIX OF THE ARC LENGTHS OF THE ELEMENTS (S-DIRECTION). DYN16864
C     IB   = FORCE ARRAY STEPPING PARAMETER, USED TO MODIFY CURRENT BLOCK OF STORAGE FOR FORCE. DYN16866
C     IELM = NUMBER OF SHELL ELEMENTS.          DYN16870
C     IHARM = MATRIX OF HARMONIC NUMBERS FOR WHICH DISPLACEMENTS AND/OR STRESSES WILL BE CALCULATED. DYN16872
C     NELEMS = TOTAL NUMBER OF ELEMENTS USED TO IDEALIZE THE STRUCTURE. DYN16876
C     NH    = TOTAL NUMBER OF HARMONICS USED IN THE DYNAMIC ANALYSIS. DYN16878
C     PHP   = DPHI/DS AT THE MIDDLE OF AN ELEMENT. DYN16880
C     TH    = MATRIX WHOSE ELEMENTS ARE THE FOURIER COEFFICIENTS OF THE CIRCUMFERENTIAL TEMPERATURE DISTRIBUTION. DYN16882
C     C                                          DYN16884
C     C                                          DYN16886
C     OUTPUT ARGUMENTS.                         DYN16888
C     FORCE = MATRIX OF GENERALIZED FORCES DUE TO EXTERNAL LOADS AND TEMPERATURES. DYN16890
C     Q    = THERMAL COEFFICIENTS USED IN CALCULATING GENERALIZED LINEAR LOADS DUE TO THERMAL EFFECTS. DYN16894
C     QQ   = GENERALIZED LINEAR LOADS DUE TO THERMAL EFFECTS. DYN16896
C     C                                          DYN16898
C     C                                          DYN16900
C     EXTERNALS.                               DYN16902
C     CALLED BY INPUT                          DYN16904
C     C                                          DYN16906
C     C                                          DYN16908
C     C                                          DYN16910
0001 SUBROUTINE TFORCE (IELM,IB)                DYN16912
0002 IMPLICIT REAL*8 (A-H,O-Z)                DYN16914
0003 COMMON /GEOM/ FNU1(50),FNU2(50),F1(50),E2(50),G(50),T(50), DYN16916
1     SINE(51),COSINE(51),SINM(50),COSM(50),R(50),PH(50), DYN16918
1     PHP(50),ARCL(50)                        DYN16920
0004 COMMON /THER/ TH(50,5,2),DTH(50,5,2),ALS(50),ALT(50) DYN16922
0005 COMMON /HARM/ NHP,IHARM(5)               DYN16924
0006 COMMON /CHALS/ AL(167),CHECK(6,8)        DYN16928
0007 COMMON /CONST/ NH,NELEMS,NNODES,NSIZE,NPRNTQ,NEQ,NEQT,N,NN,NHNS, DYN16930
1     DT2,NPRNTL,NPRNTF,IDELEF,IDCOE         DYN16934
0008 COMMON /QS/ QN(1020),QN1(1020),FORCE(2040),QP(1020),QP1(1020), DYN16940
1     QN2(1020)                               DYN16950
0009 COMMON /QUES/ Q(8),QQ(8)                 DYN16960
0010 DIMENSION CES(4)                         DYN16970
C1  LINEAR THERMAL LOADS                      DYN16972
0011 PI=3.14159265                            DYN17010
0012 CES(1)=PI*E1(IELM)*T(IELM)/(1.-FNU1(IELM)*FNU2(IELM))*(ALS(IELM)+ DYN17020
1     FNU1(IELM)*ALT(IELM))                   DYN17030
0013 CES(2)=PI*E2(IELM)*T(IELM)/(1.-FNU1(IELM)*FNU2(IELM))*(ALT(IELM)+ DYN17040

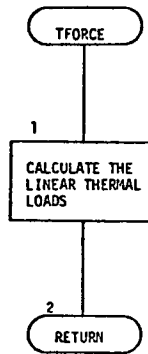
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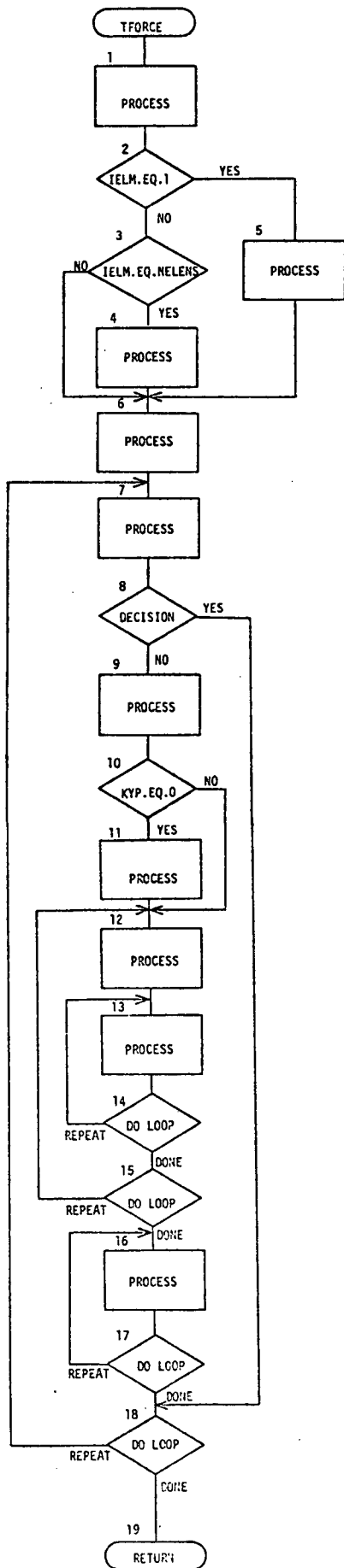
0014      1      FNU2(IELM)*ALS(IELM)                                DYN17050
          CES(3)=PI*E1(IELM)*T(IELM)**3/(12.*(1.-FNU1(IELM)*FNU2(IELM)))* DYN17060
          1      {ALS(IELM)+FNU1(IELM)*ALT(IELM)}                    DYN17070
0015      CES(4)=PI*E2(IELM)*T(IELM)**3/(12.*(1.-FNU1(IELM)*FNU2(IELM)))* DYN17080
          1      {ALT(IELM)+FNU2(IELM)*ALS(IELM)}                    DYN17090
          C      DYN17100
          C      CALCULATE PHPP(IELM)                                DYN17110
          C      DYN17120
0016      IF (IELM.EQ.1) GO TO 10                                    DYN17130
0017      IF (IELM.EQ.NELEMS) GO TO 20                              DYN17140
0018      PHPP1=(PHP(IELM)-PHP(IELM-1))/(ARCL(IELM)+ARCL(IELM-1))*2. DYN17150
0019      PHPP2=(PHP(IELM+1)-PHP(IELM))/(ARCL(IELM)+ARCL(IELM+1))*2. DYN17160
0020      PHPP=(PHPP1+PHPP2)/2.                                       DYN17170
0021      GO TO 30                                                    DYN17180
0022      10 PHPP=(PHP(IELM+1)-PHP(IELM))/(ARCL(IELM)+ARCL(IELM+1))*2. DYN17190
0023      GO TO 30                                                    DYN17200
0024      20 PHPP=(PHP(IELM)-PHP(IELM-1))/(ARCL(IELM)+ARCL(IELM-1))*2. DYN17210
0025      30 CONTINUE                                                DYN17220
0026      IBP1=IB+1                                                  DYN17230
          C      DYN17238
0027      DO 90 IH=1,NH                                             DYN17240
0028      KYP=IHARM(IH)                                             DYN17250
0029      YKP=KYP                                                    DYN17260
0030      IF (TH(IELM,IH,IBP1).EQ.0.AND.DTH(IELM,IH,IBP1).EQ.0) GO TO 90 DYN17270
0031      Q(1)=(-CES(1)*AL(4)+CES(2)*AL(17))*TH(IELM,IH,IBP1)+YKP**2* DYN17280
          1      CES(4)*AL(44)*DTH(IELM,IH,IBP1)                    DYN17290
0032      Q(2)=(-CES(1)*AL(5)+CES(2)*AL(18))*TH(IELM,IH,IBP1)+(YKP**2* DYN17300
          1      AL(45)-AL(21))*CES(4)*DTH(IELM,IH,IBP1)          DYN17310
0033      Q(3)=(-CES(1)*AL(6)+CES(2)*AL(19))*TH(IELM,IH,IBP1)+(-2.* DYN17320
          1      CES(3)*AL(1)+CES(4)*(YKP**2*AL(46)-2.*AL(22)))* DYN17330
          1      DTH(IELM,IH,IBP1)                                    DYN17331
0034      Q(4)=(-CES(1)*AL(7)+CES(2)*AL(20))*TH(IELM,IH,IBP1)+(-6.* DYN17340
          1      CES(3)*AL(2)+CES(4)*(YKP**2*AL(47)-3.*AL(23)))* DYN17350
          1      DTH(IELM,IH,IBP1)                                    DYN17351
0035      Q(5)=CES(2)*AL(21)*TH(IELM,IH,IBP1)-(CES(3)*AL(1)*PHPP+CES(4)* DYN17360
          1      AL(37))*DTH(IELM,IH,IBP1)                          DYN17370
0036      Q(6)=(CES(1)*AL(1)+CES(2)*AL(22))*TH(IELM,IH,IBP1)-(CES(3)* DYN17380
          1      (AL(4)+AL(2)*PHPP)+CES(4)*AL(38))*DTH(IELM,IH,IBP1) DYN17390
0037      Q(7)=CES(2)*AL(15)*YKP*TH(IELM,IH,IBP1)+CES(4)*AL(49)*YKP* DYN17400
          1      DTH(IELM,IH,IBP1)                                    DYN17410
0038      Q(8)=CES(2)*AL(16)*YKP*TH(IELM,IH,IBP1)+CES(4)*AL(50)*YKP* DYN17420
          1      DTH(IELM,IH,IBP1)                                    DYN17430
0039      IF (KYP.EQ.0) GO TO 40                                       DYN17440
0040      GO TO 60                                                    DYN17450
          C      DYN17458
0041      40 DO 50 J=1,8                                             DYN17460
0042      Q(J)=2.*Q(J)                                               DYN17470
0043      50 CONTINUE                                                DYN17472

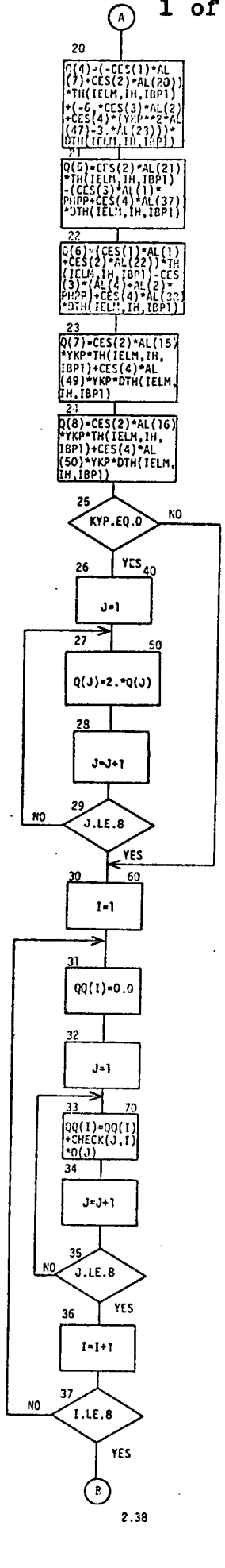
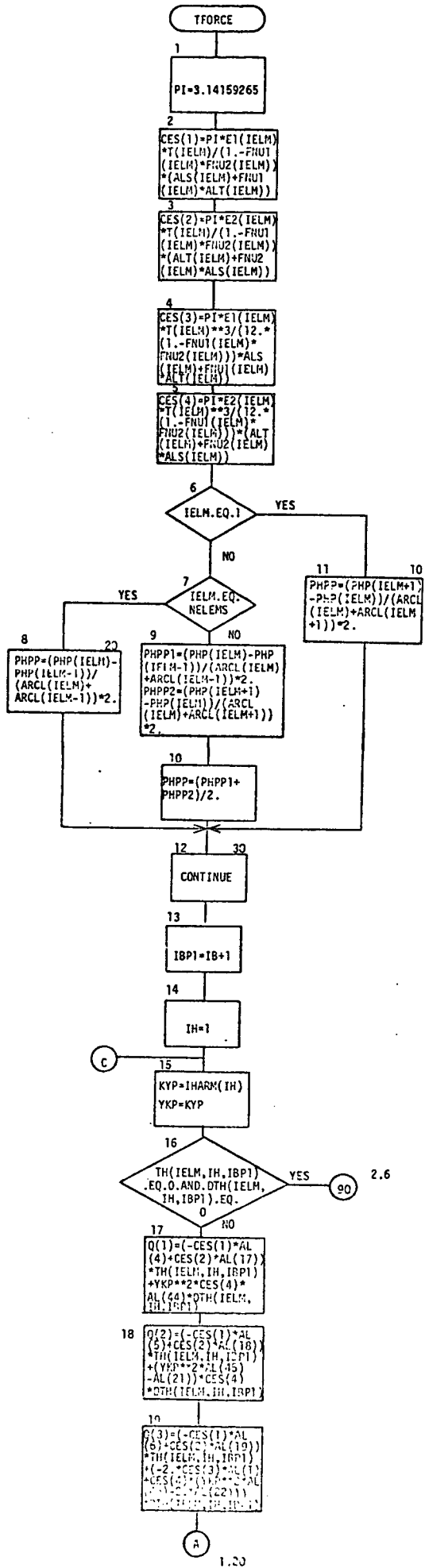
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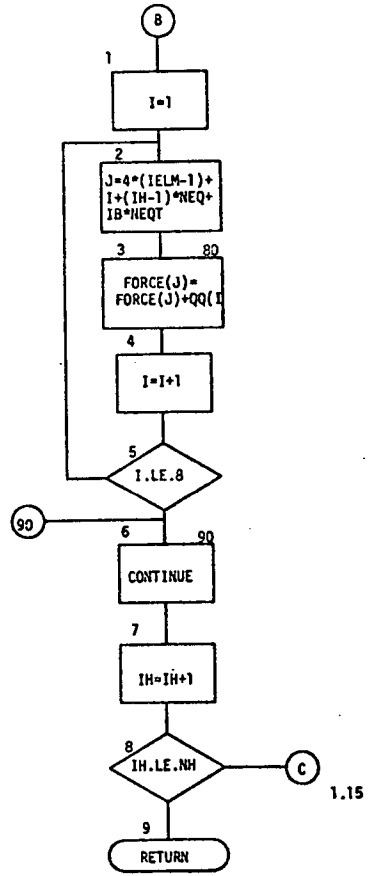
	C		DYN17475
	C		DYN17478
0044		60 DO 70 I=1,8	DYN17480
0045		QQ(I)=0.0	DYN17490
	C		DYN17498
0046		DO 70 J=1,8	DYN17500
0047		QQ(I)=QQ(I)+CHECK(J,I)*Q(J)	DYN17510
0048		70 CONTINUE	DYN17512
	C		DYN17515
	C		DYN17518
0049		DO 80 I=1,8	DYN17520
0050		J=4*(IELM-1)+I+(IH-1)*NEQ+IB*NEQ	DYN17530
0051		FORCE(J)=FORCE(J)+QQ(I)	DYN17540
0052		80 CONTINUE	DYN17542
	C		DYN17545
0053		90 CONTINUE	DYN17550
	C		DYN17553
0054		RETURN	DYN17560
0055		END	DYN17570











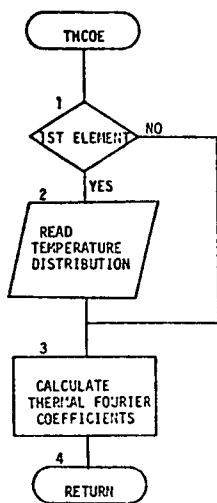
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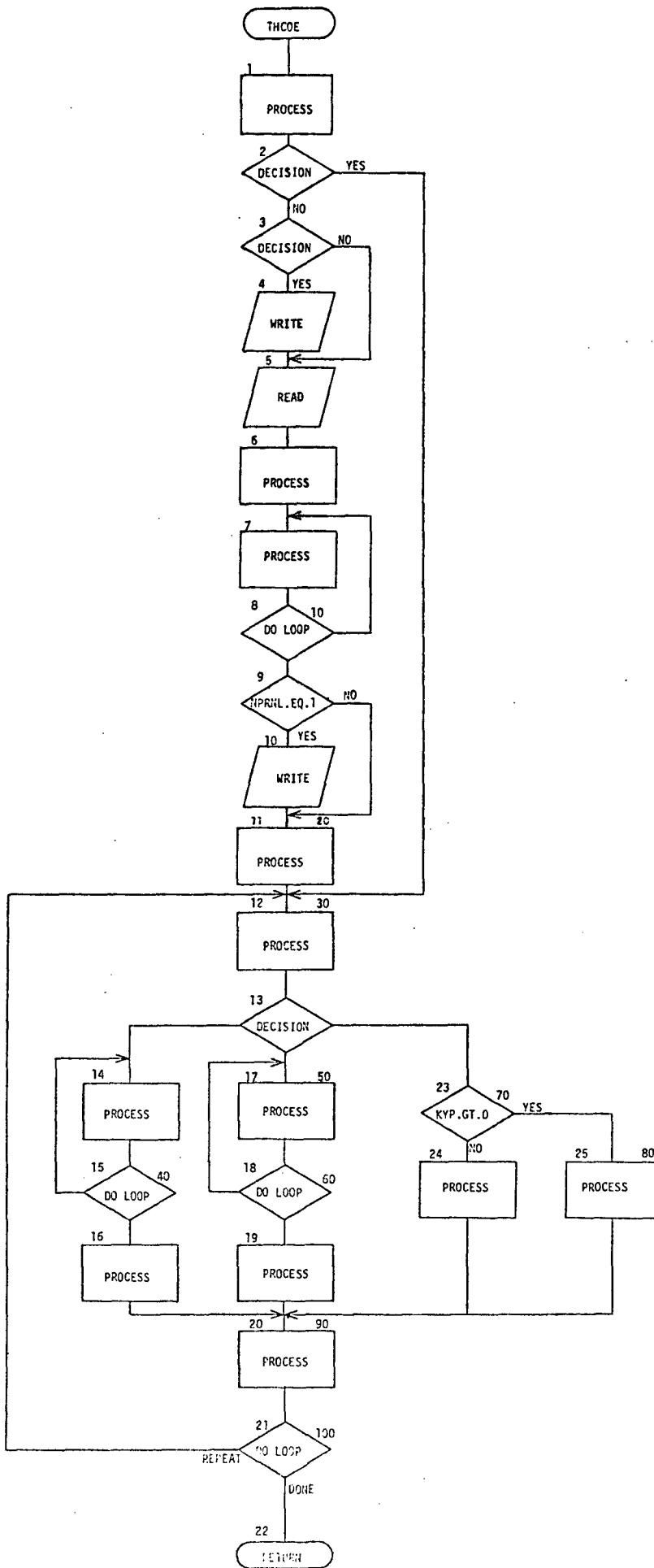
C(THCOE) DYN16102
C DYN16104
C DESCRIPTION - TO READ IN THE TEMPERATURE AND TEMPERATURE DYN16106
C GRADIENTS FOR UTILIZATION IN COMPUTING THE THERMAL DYN16108
C FOURIER COEFFICIENTS. DYN16110
C DYN16112
C INPUT ARGUMENTS. DYN16114
C IB = FORCE ARRAY STEPPING PARAMETER, USED TO MODIFY CURRENT DYN16116
C BLOCK OF STORAGE FOR FORCE. DYN16118
C IELM = NUMBER OF SHELL ELEMENTS. DYN16120
C IHARM = MATRIX OF HARMONIC NUMBERS FOR WHICH DISPLACEMENTS DYN16122
C AND/OR STRESSES WILL BE CALCULATED. DYN16124
C NH = TOTAL NUMBER OF HARMONICS USED IN THE DYNAMIC ANALYSIS. DYN16126
C DYN16128
C OUTPUT ARGUMENTS. DYN16130
C DTH = MATRIX OF FOURIER COEFFICIENTS FOR THE CIRCUMFERENTIAL DYN16132
C TEMPERATURE GRADIENT DISTRIBUTION. DYN16134
C TH = MATRIX WHOSE ELEMENTS ARE THE FOURIER COEFFICIENTS DYN16136
C OF THE CIRCUMFERENTIAL TEMPERATURE DISTRIBUTION. DYN16138
C DYN16140
C EXTERNALS. DYN16142
C CALLED BY DYN16144
C INPUT DYN16146
C DYN16148
C DYN16150
0001 SUBROUTINE THCOE (IELM,IB) DYN16152
0002 IMPLICIT REAL*8 (A-H,O-Z) DYN16154
0003 COMMON /CONST/ NH,NELEMS,NNODES,NSIZE,NPRNTQ,NEQ,NEQT,N,NN,NHNS, DYN16156
1 DT2,NPRNTL,NPRNTF,IDEFL,IDCOE DYN16158
0004 COMMON /FRCE/ P(74),R(74),S(74),THETB(74) DYN16160
0005 COMMON /THER/ TH(50,5,2),DTH(50,5,2),ALS(50),ALT(50) DYN16162
0006 COMMON /HARM/ NHP,IHARM(5) DYN16170
0007 COMMON /TAPES/ NT,ND,NS DYN16220
0008 IF (IELM.EQ.1) IELM2=0 DYN16230
0009 PI=3.14159265 DYN16232
C1 FIRST ELEMENT//NO(3C) DYN16240
0010 IF (IELM.LE.IELM2.AND.IELM.NE.1) GO TO 30 DYN16250
0011 IF (NPRNTL.EQ.1.AND.IELM.EQ.1) WRITE (6,110) DYN16260
C DYN16262
C1S(IO) READ TEMPERATURE DISTRIBUTION DYN16280
C DYN16290
C READ INPUT DATA FOR CARD TYPE IX - D - 1 DYN16300
0012 READ (ND,120) IELM1,IELM2,NDP,(THETB(I),P(I),R(I),I=1,NDP) DYN16310
0013 NDP2=2*NDP+1 DYN16318
C DYN16320
0014 DO 10 IF=1,NDP DYN16330
0015 ANG=360.0-THETB(NDP) DYN16340
0016 KEY=NDP2-IF DYN16350
0017 THETB(KEY)=ANG

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0018		P(KEY)=P(IF)	DYN16360
0019		R(KEY)=R(IF)	DYN16370
0020	10	CONTINUE	DYN16380
	C		DYN16383
0021		IF (NPRNTL.EQ.1) WRITE (6,130) IELM1,IELM2,(P(I),R(I),THETB(I),	DYN16390
	1	THETB(I+1),I=1,NDP)	DYN16400
0022		ND2=2*NDP	DYN16410
	C		DYN16418
0023		DO 20 IDP=1,ND2	DYN16420
0024		THETB(IDP)=THETB(IDP)/57.2957795	DYN16430
0025	20	CONTINUE	DYN16432
	C		DYN16435
0026	30	CONTINUE	DYN16440
	C1	CALCULATE THERMAL FOURIER COEFFICIENTS	DYN16442
	C		DYN16448
0027		DO 100 IH=1,NH	DYN16450
0028		KYP=IHARM(IH)	DYN16460
0029		YKP=KYP	DYN16470
0030		PINT=0.0	DYN16480
0031		RINT=0.0	DYN16490
0032		IF (NDP.EQ.1) GO TO 70	DYN16500
0033		IF (KYP.GT.0) GO TO 50	DYN16510
	C		DYN16518
0034		DO 40 I=1,NDP	DYN16520
0035		PINT=PINT+P(I)*(THETB(I+1)-THETB(I))	DYN16530
0036		RINT=RINT+R(I)*(THETB(I+1)-THETB(I))	DYN16540
0037	40	CONTINUE	DYN16542
	C		DYN16545
0038		PINT=PINT/(2.*PI)	DYN16550
0039		RINT=RINT/(2.*PI)	DYN16560
0040		GO TO 90	DYN16570
	C		DYN16578
0041	50	DO 60 I=1,NDP	DYN16580
0042		X1=THETB(I)*YKP	DYN16590
0043		X2=THETB(I+1)*YKP	DYN16600
0044		PINT=PINT+P(I)*(DSIN(X2)-DSIN(X1))/YKP	DYN16610
0045		RINT=RINT+R(I)*(DSIN(X2)-DSIN(X1))/YKP	DYN16620
0046	60	CONTINUE	DYN16622
	C		DYN16625
0047		PINT=PINT/PI	DYN16630
0048		RINT=RINT/PI	DYN16640
0049		GO TO 90	DYN16650
0050	70	IF (KYP.GT.0) GO TO 80	DYN16660
0051		PINT=P(I)	DYN16670
0052		RINT=R(I)	DYN16680
0053		GO TO 90	DYN16690
0054	80	PINT=0.0	DYN16700
0055		RINT=0.0	DYN16710

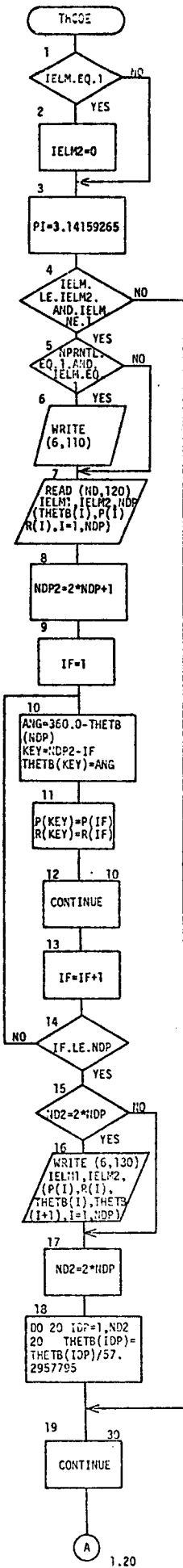
0056	90	IBP1=IB+1	DYN16720
0057		TH(IELM,IH,IBP1)=PINT	DYN16730
0058		DTH(IELM,IH,IBP1)=RINT	DYN16740
0059		100 CONTINUE	DYN16750
	C		DYN16753
0060		RETURN	DYN16760
	C		DYN16770
0061		110 FORMAT (1H1,31X,4HTEMPERATURES AND THERMAL GRADIENTS ON,	DYN16780
		1 10H STRUCTURE//	DYN16782
		1 27X,11HTEMPERATURE,10X,16HTHERMAL GRADIENT,10X,	DYN16790
		2 29HFROM THETA TO THETA (DEGREES)//	DYN16800
0062		120 FORMAT (3I5/(3F10.0))	DYN16810
0063		130 FORMAT (/,60X,11HELEMENT NO.,I3,1H-,I2,//	DYN16820
		1 (28X,F9.3,15X,F10.3,16X,F7.2,2X,F7.2))	DYN16830
0064		END	DYN16840





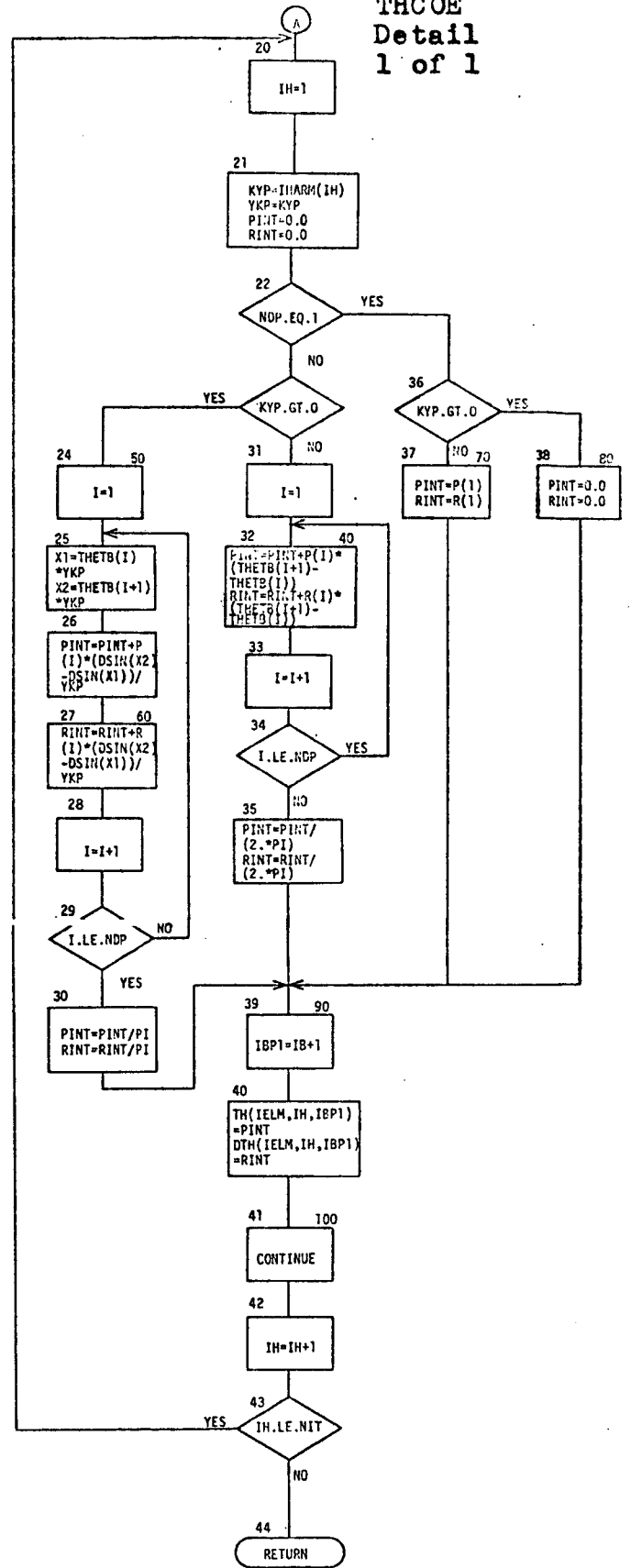


J-3.132



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THCOE  
Detail  
1 of 1



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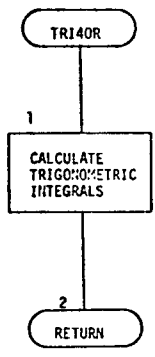
CE(TRI4OR)                                DYN17582
C                                           DYN17584
C      DESCRIPTION - TO CALCULATE THE VARIOUS TRIGONOMETRIC      DYN17586
C      INTEGRALS REQUIRED IN THE CALCULATIONS OF THE              DYN17588
C      GENERALIZED NONLINEAR LOADS.                             DYN17590
C                                                                DYN17592
C INPUT ARGUMENTS.                                           DYN17594
C      IHARM = MATRIX OF HARMONIC NUMBERS FOR WHICH DISPLACEMENTS DYN17596
C      AND/OR STRESSES WILL BE CALCULATED.                    DYN17598
C      NH    = TOTAL NUMBER OF HARMONICS USED IN THE DYNAMIC ANALYSIS. DYN17600
C                                                                DYN17602
C OUTPUT ARGUMENTS.                                         DYN17604
C      CCC   = MATRIX CONTAINING INTEGRALS FROM 0 TO 2*PI OF    DYN17606
C      COS(I*THETA) * COS(J*THETA) * COS(K*THETA) * DTHETA.    DYN17608
C      CCCC  = MATRIX CONTAINING INTEGRALS FROM 0 TO 2*PI OF    DYN17610
C      COS(I*THETA) * COS(J*THETA) * COS(K*THETA) *          DYN17612
C      COS(L*THETA) * DTHETA.                                    DYN17614
C      CSS   = MATRIX CONTAINING INTEGRALS FROM 0 TO 2*PI OF    DYN17616
C      COS(I*THETA) * SIN(J*THETA) * SIN(K*THETA) * DTHETA.    DYN17618
C      SSC   = MATRIX CONTAINING INTEGRALS FROM 0 TO 2*PI OF    DYN17620
C      SIN(I*THETA) * SIN(J*THETA) * COS(K*THETA) * DTHETA.    DYN17622
C      SSCC  = MATRIX CONTAINING INTEGRALS FROM 0 TO 2*PI OF    DYN17624
C      SIN(I*THETA) * SIN(J*THETA) * COS(K*THETA) *          DYN17626
C      COS(L*THETA) * DTHETA.                                    DYN17628
C      SSSS  = MATRIX CONTAINING INTEGRALS FROM 0 TO 2*PI OF    DYN17630
C      SIN(I*THETA) * SIN(J*THETA) * SIN(K*THETA) *          DYN17632
C      SIN(L*THETA) * DTHETA.                                    DYN17634
C                                                                DYN17636
C EXTERNALS.                                               DYN17638
C      CALLED BY                                           DYN17640
C      INPUT                                               DYN17642
C                                                                DYN17644
0001      SUBROUTINE TRI4OR                                DYN17646
0002      IMPLICIT REAL*8 (A-H,O-Z)                        DYN17648
0003      COMMON /CS/ CCC(125),SSC(125),CSS(125)           DYN17650
0004      COMMON /CS4/ CCCC(625),SSSS(625),SSCC(625),SCCS(625) DYN17652
0005      COMMON /CONST/ NH,NELEMS,NNODES,NSIZE,NPRNTQ,NEQ,NEQT,N,NN,NHNS, DYN17654
0006      1      DT2,NPRNTL,NPRNTF,IDELF,IDCOE             DYN17656
0007      COMMON /HARM/ NHP,IHARM(5)                       DYN17658
0008      C1      CALCULATE TRIGONOMETRIC INTEGRALS        DYN17660
0009      PIO2=1.570796                                     DYN17690
0010      ITH=0                                             DYN17700
0011      C                                               DYN17708
0012      DO 10 M=1,NH                                     DYN17710
0013          K=IHARM(M)                                    DYN17720
0014      C                                               DYN17728
0015          DO 10 I=1,NH                                  DYN17730
0016              II=IHARM(I)                               DYN17740
    
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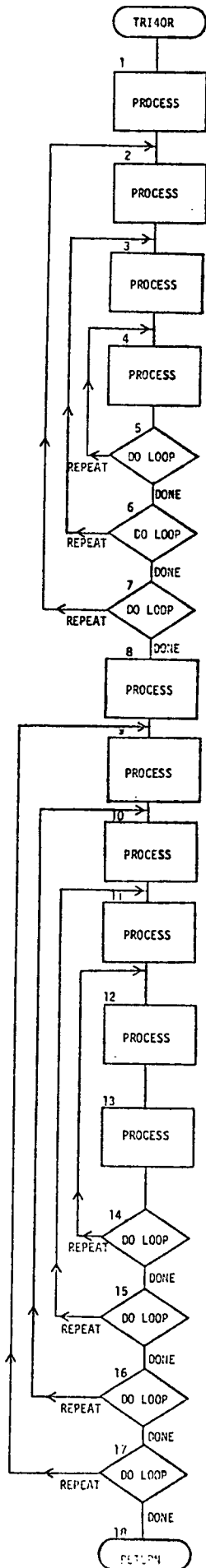
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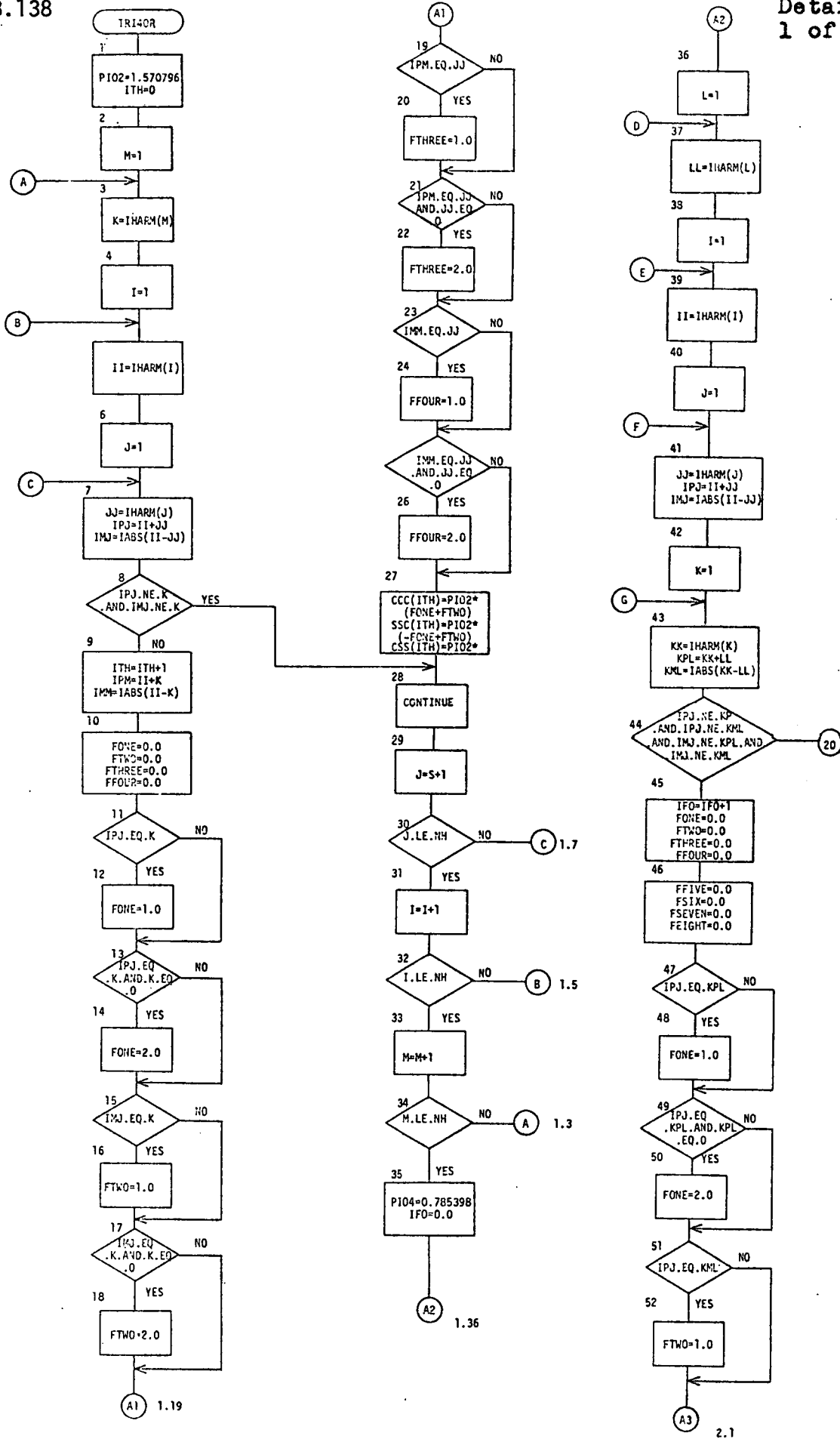
C
0013      DO 10 J=1,NH                DYN17748
0014          JJ=IHARM(J)            DYN17750
0015          IPJ=II+JJ              DYN17760
0016          IMJ=IABS(II-JJ)        DYN17770
0017          IF (IPJ.NE.K.AND.IMJ.NE.K) GO TO 10 DYN17780
0018          ITH=ITH+1              DYN17790
0019          IPM=II+K                DYN17800
0020          IMM=IABS(II-K)          DYN17810
0021          FONE=0.0                DYN17820
0022          FTWO=0.0                DYN17830
0023          FTHREE=0.0              DYN17840
0024          FFOUR=0.0               DYN17850
0025          IF (IPJ.EQ.K) FONE=1.0 DYN17860
0026          IF (IPJ.FQ.K.AND.K.EQ.0) FONE=2.0 DYN17870
0027          IF (IMJ.EQ.K) FTWO=1.0 DYN17880
0028          IF (IMJ.EQ.K.AND.K.EQ.0) FTWO=2.0 DYN17890
0029          IF (IPM.EQ.JJ) FTHREE=1.0 DYN17900
0030          IF (IPM.EQ.JJ.AND.JJ.EQ.0) FTHREE=2.0 DYN17910
0031          IF (IMM.EQ.JJ) FFOUR=1.0 DYN17920
0032          IF (IMM.EQ.JJ.AND.JJ.EQ.0) FFOUR=2.0 DYN17930
0033          CCC(ITH)=PIO2*(FONE+FTWO) DYN17940
0034          SSC(ITH)=PIO2*(-FONE+FTWO) DYN17950
0035          CSS(ITH)=PIO2*(-FTHREE+FFOUR) DYN17960
0036      10 CONTINUE                DYN17970
C
0037      PID4=0.785398              DYN17980
0038      IFO=0.0                    DYN17983
C
0039      DO 20 L=1,NH                DYN17990
0040          LL=IHARM(L)             DYN18000
C
0041      DO 20 I=1,NH                DYN18008
0042          II=IHARM(I)            DYN18010
C
0043      DO 20 J=1,NH                DYN18020
0044          JJ=IHARM(J)            DYN18028
0045          IPJ=II+JJ              DYN18030
0046          IMJ=IABS(II-JJ)        DYN18040
C
0047      DO 20 K=1,NH                DYN18048
0048          KK=IHARM(K)            DYN18050
0049          KPL=KK+LL              DYN18060
0050          KML=IABS(KK-LL)        DYN18070
0051          IF (IPJ.NE.KPL.AND.IPJ.NE.KML.AND.IMJ. DYN18080
1          NE.KPL.AND.IMJ.NE.KML) GO TO 20 DYN18088
0052          IFO=IFO+1              DYN18090
0053          FONE=0.0                DYN18100

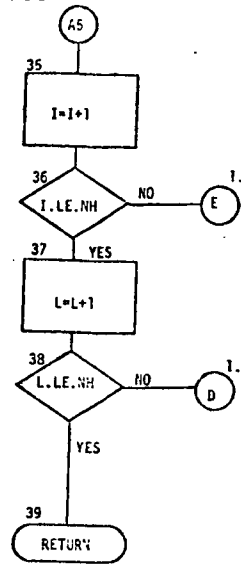
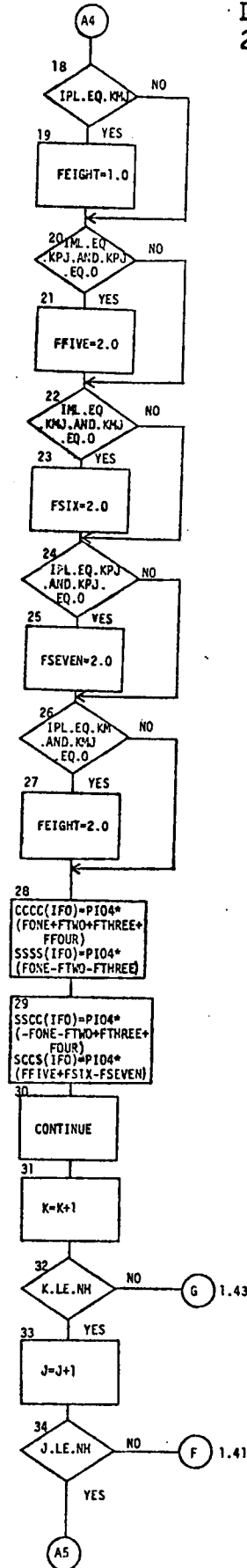
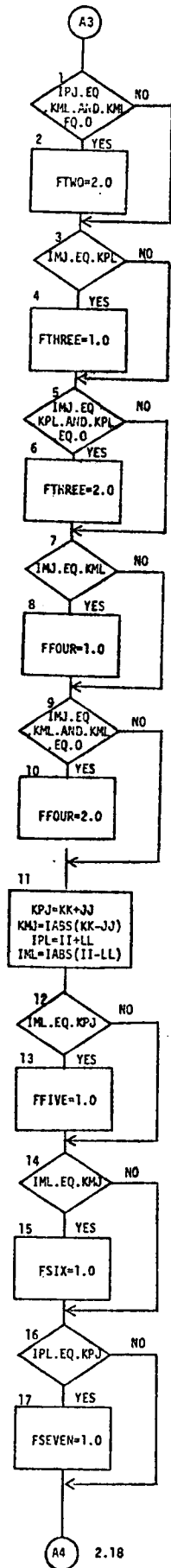
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0054	FTWO=0.0	DYN18170
0055	FTHREE=0.0	DYN18180
0056	FFOUR=0.0	DYN18190
0057	FFIVE=0.0	DYN18200
0058	FSIX=0.0	DYN18210
0059	FSEVEN=0.0	DYN18220
0060	FEIGHT=0.0	DYN18230
0061	IF (IPJ.EQ.KPL) FONE=1.0	DYN18240
0062	IF (IPJ.EQ.KPL.AND.KPL.EQ.0) FONE=2.0	DYN18250
0063	IF (IPJ.EQ.KML) FTWO=1.0	DYN18260
0064	IF (IPJ.EQ.KML.AND.KML.EQ.0) FTWO=2.0	DYN18270
0065	IF (IMJ.EQ.KPL) FTHREE=1.0	DYN18280
0066	IF (IMJ.EQ.KPL.AND.KPL.EQ.0) FTHREE=2.0	DYN18290
0067	IF (IMJ.EQ.KML) FFOUR=1.0	DYN18300
0068	IF (IMJ.EQ.KML.AND.KML.EQ.0) FFOUR=2.0	DYN18310
0069	KPJ=KK+JJ	DYN18320
0070	KMJ=IABS(KK-JJ)	DYN18330
0071	IPL=II+LL	DYN18340
0072	IML=IABS(II-LL)	DYN18350
0073	IF (IML.EQ.KPJ) FFIVE=1.0	DYN18360
0074	IF (IML.EQ.KMJ) FSIX=1.0	DYN18370
0075	IF (IPL.EQ.KPJ) FSEVEN=1.0	DYN18380
0076	IF (IPL.EQ.KMJ) FEIGHT=1.0	DYN18390
0077	IF (IML.EQ.KPJ.AND.KPJ.EQ.0) FFIVE=2.0	DYN18400
0078	IF (IML.EQ.KMJ.AND.KMJ.EQ.0) FSIX=2.0	DYN18410
0079	IF (IPL.EQ.KPJ.AND.KPJ.EQ.0) FSEVEN=2.0	DYN18420
0080	IF (IPL.EQ.KMJ.AND.KMJ.EQ.0) FEIGHT=2.0	DYN18430
0081	CCCC(IF0)=PI04*(FONE+FTWO+FTHREE+FFOUR)	DYN18440
0082	SSSS(IF0)=PI04*(FONE-FTWO-FTHREE+FFOUR)	DYN18450
0083	SSCC(IF0)=PI04*(-FONE-FTWO+FTHREE+FFOUR)	DYN18460
0084	SCCS(IF0)=PI04*(FFIVE+FSIX-FSEVEN-FEIGHT)	DYN18470
0085		DYN18480
	20 CONTINUE	DYN18483
	C	DYN18490
0086	RETURN	DYN18500
0087	END	











## VARIABLE CROSS REFERENCE

The variable cross reference listing gives an alphabetical listing of all variables from each routine along with the type of each variable, the dimension value for variable arrays, the statement number referencing the variable, and a corresponding letter value for each statement reference. The letter values for each reference are one of the following:

- U - this indicates that the variable is simply being used in this particular statement reference. No values are being assigned to the variable in this particular reference.
- D - a reference with a corresponding letter D indicates that the variable is defined in this statement reference. Examples of statements in which variables could be defined are COMMON, DIMENSION, all type statements, and subroutine definitions.
- S - this indicates that the variable is set or given a value as in an assignment statement, input statement, etc...

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P - the letter P stands for parameter and indicates that the variable appears in the argument list of a subroutine CALL statement.

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
A	REAL*8		204	10850,U 10870,D 10950,S 11040,S 11040,S 11050,U 11050,S 11060,U 11060,S 11070,U 11070,U 11180,U 11180,S 11190,U 11190,S 11210,S 11210,U 11310,U 11410,S 11460,S 11460,U 11460,U 11470,U 11470,U 11470,S 11470,U 11470,U 11480,S 11480,U 11480,U 11480,U 11490,U 11490,U 11490,S 11490,U 11500,S 11500,U 11500,U 11510,U 11510,S 11510,U 11520,U 11520,S 11520,U 11520,U 11520,U 11520,U 11520,U 11520,U 11520,U 11530,U 11530,U 11530,U 11530,U 11530,S 11540,U 11540,S 11540,U 11550,S 11550,U 11550,U 11560,U 11560,S 11560,U 11570,S 11570,U 11570,U 11570,U 11570,U 11570,U 11570,U 11570,U 11570,U 11570,U 11570,U 11580,U 11590,U 11590,U 11600,U 11600,U 11610,U 11650,U 11660,U 11670,U 11680,U 11690,U 11700,U 11720,U 11730,U 11740,U 11750,U 11760,U 11770,U 11780,U 11790,U 11800,U 11810,U 11820,U 11830,U 11840,U 11850,U 11860,U 11870,U 11880,U 11890,U 11900,U 11910,U 11920,U 11930,U 11940,U 11950,U 11960,U 11970,U 11980,U 11990,U 12000,U 12010,U 12020,U 12030,U 12040,U 12050,U 12060,U 12070,U 12600,S 12610,S 12620,S 12630,S 12640,S 12650,S 12660,S 12690,S 12700,S 12710,S 12720,S 12730,S 12740,S 12750,S 12760,S 12770,S 12780,S 12790,S 12800,S 12810,S 12820,S 12830,S 12840,S 12850,S 12860,S 12870,S 12880,S 12890,S 12900,S 12910,S 12920,S 12930,S 12940,S 12950,S 12960,S 12970,S 12980,S 12990,S 13000,S 13010,S 13020,S 13030,S 13060,U 13070,U 13070,U 13080,U 13090,U 13080,U 13120,U 13120,U 13120,U 13120,U 13120,U 13130,U 13130,U 13130,U 13130,U 13150,U 13150,U 13150,U 13160,U 13160,U 13160,U 13170,U 13170,U 13170,U 13180,U 13180,U 13180,U 13180,U 13240,U 13250,U 13250,U 13260,U 13260,U 13270,U 13270,U 13270,U 13270,U 13270,U 13320,U 13320,U 13320,U 13320,U 13330,U 13340,U 13340,U 13340,U 13340,U 13350,U 13350,U 13360,U 13360,U 13360,U 13360,U 13370,U 13370,U 13370,U 13380,U 13380,U 13380,U 13380,U 13390,U 13390,U 13390,U 13390,U 13410,U 13410,U 13410,U 13410,U 13420,U 13430,U 13430,U 13430,U 13430,U 13440,U 13440,U 13450,U 13450,U 13450,U 13450,U 13460,U 13460,U 13460,U 13470,U 13470,U 13470,U 13470,U 13480,U 13480,U 13480,U 13480,U 13550,U
AAV	REAL*8			14590,S 14600,U 14620,U
AL	REAL*8		167	930,D 2650,S 2660,U 4310,S 4490,S 14900,D 15640,U 15650,U 15660,U 15670,U 15680,U 15690,U 15700,U 15710,U 15790,U 15800,U 15810,U 15820,U 15830,U 15840,U 15850,U 15860,U 16910,D 17280,U 17280,U 17290,U 17300,U 17300,U 17310,U 17310,U 17320,U 17320,U 17320,U 17330,U 17330,U 17340,U 17340,U 17340,U 17350,U 17350,U 17360,U 17360,U 17360,U 17380,U 17380,U 17380,U 17380,U 17390,U 17390,U 17400,U 17420,U 17420,U
ALPHK	REAL*8			4320,P 14870,U
ALS	REAL*8		50	1080,D 2890,S 2940,S 3020,U 7170,D 8820,U 8830,U 13710,D 14090,U 14150,U 16150,D 16890,D 17020,U 17050,U 17060,U 17090,U
ALSI1	REAL*8			2870,S 2890,U
ALT	REAL*8		50	1080,D 2900,S 2950,S 3020,U 7170,D 8820,U 8830,U 13710,D 14100,U 14180,U 16150,D 16890,D 17030,U 17040,U 17070,U 17080,U
ALTI1	REAL*8			2870,S 2900,U
AM1	REAL*8			11810,S 12230,U 12310,U 12330,U 12400,U 12430,U 12510,U 12550,U 12770,U
AM10	REAL*8			11650,S 11720,S 12200,U 12230,U 12280,U 12330,U 12370,U 12430,U 12480,U 12550,U 12630,U 12680,U
AM2	REAL*8			11800,S 12160,U 12170,U 12180,U 12190,U 12760,U

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
AM3	REAL*8			11790,S 12160,U 12170,U 12180,U 12190,U 12750,U
AM4	REAL*8			11780,S 12160,U 12170,U 12180,U 12190,U 12740,U
AM5	REAL*8			11700,S 11770,S 12220,U 12230,U 12300,U 12330,U 12390,U 12430,U 12500,U 12550,U
AM6	REAL*8			11690,S 11760,S 12120,U 12130,U 12140,U 12150,U 12640,U 12720,U
AM7	REAL*8			11680,S 11750,S 12120,U 12130,U 12140,U 12150,U 12630,U 12710,U
AM8	REAL*8			11670,S 11740,S 12210,U 12230,U 12290,U 12330,U 12380,U 12430,U 12490,U 12550,U
AM9	REAL*8			11660,S 11730,S 12080,U 12090,U 12100,U 12110,U 12610,U 12690,U
ANG	REAL*8			15200,S 15220,U 16330,S 16350,U
ARCL	REAL*8	50		1030,D 2720,S 2740,S 3030,U 6610,D 7140,D 13650,D 14140,U 14210,U 14210,U 14590,U 14590,U 16880,D 17150,U 17150,U 17160,U 17160,U 17190,U 17190,U 17210,U 17210,U
ARCLI	REAL*8			7380,S 7440,U 7450,U 7460,U 7470,U 7480,U 7490,U
ARL	REAL*8			7330,S 7340,U 7350,U 7380,U 7900,U 8670,U
A0	REAL*8			11820,S 12080,U 12120,U 12160,U 12200,S 12200,U 12230,U 12230,U 12240,U 12250,U 12260,U 12270,U 12780,U
A1	REAL*8			11830,S 12080,S 12080,U 12120,U 12160,U 12210,S 12210,S 12230,U 12230,U 12240,U 12250,U 12260,U 12270,U 12790,U
A10	REAL*8			11920,S 12330,U 12330,S 12420,U 12440,U 12530,U 12560,U 12880,U
A11	REAL*8			11930,S 12100,U 12140,U 12180,U 12250,U 12340,U 12370,U 12370,S 12430,U 12430,U 12450,U 12460,U 12890,U
A12	REAL*8			11940,S 12100,U 12100,S 12140,U 12180,U 12250,U 12340,U 12380,S 12380,U 12430,U 12430,U 12450,U 12460,U 12900,U
A13	REAL*8			11950,S 12140,U 12140,S 12180,U 12250,U 12340,U 12390,S 12390,U 12430,U 12430,U 12450,U 12460,U 12910,U
A14	REAL*8			11960,S 12180,U 12180,S 12250,U 12340,U 12400,S 12400,U 12430,U 12430,U 12450,U 12460,U 12920,U
A15	REAL*8			11970,S 12250,U 12250,S 12340,U 12410,U 12410,S 12430,U 12430,U 12450,U 12460,U 12930,U
A16	REAL*8			11980,S 12340,U 12340,S 12420,S 12420,U 12440,U 12440,U 12450,U 12470,U 12940,U
A17	REAL*8			11990,S 12430,S 12430,U 12540,U 12560,U 12950,U

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VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
A18	REAL*8			12000,S 12110,U 12150,U 12190,U 12260,U 12350,U 12450,U 12480,S 12480,U 12550,U 12550,U 12570,U 12960,U
A19	REAL*8			12010,S 12110,S 12110,U 12150,U 12190,U 12260,U 12350,U 12450,U 12490,S 12490,U 12550,U 12550,U 12570,U 12970,U
A2	REAL*8			11840,S 12120,S 12120,U 12160,U 12220,U 12220,S 12230,U 12230,U 12240,U 12250,U 12260,U 12270,U 12800,U
A20	REAL*8			12020,S 12150,U 12150,S 12190,U 12260,U 12350,U 12450,U 12500,S 12500,U 12550,U 12550,U 12570,U 12980,U
A21	REAL*8			12030,S 12190,S 12190,U 12260,U 12350,U 12450,U 12510,S 12510,U 12550,U 12550,U 12570,U 12990,U
A22	REAL*8			12040,S 12260,U 12260,S 12350,U 12450,U 12520,S 12520,U 12550,U 12550,U 12570,U 13000,U
A23	REAL*8			12050,S 12350,S 12350,U 12450,U 12530,S 12530,U 12560,U 12560,U 12580,U 13010,U
A24	REAL*8			12060,S 12450,S 12450,U 12540,U 12540,S 12560,U 12560,U 12580,U 13020,U
A25	REAL*8			12070,S 12550,S 12550,U 13030,U
A3	REAL*8			11850,S 12160,S 12160,U 12230,U 12230,U 12240,U 12250,U 12260,U 12270,U 12810,U
A4	REAL*8			11860,S 12230,U 12230,S 12320,U 12330,U 12410,U 12430,U 12520,U 12550,U 12820,U
A5	REAL*8			11870,S 12090,U 12130,U 12170,U 12240,U 12280,S 12280,U 12330,U 12330,U 12340,U 12350,U 12360,U 12830,U
A6	REAL*8			11880,S 12090,S 12090,U 12130,U 12170,U 12240,U 12290,S 12290,U 12330,U 12330,U 12340,U 12350,U 12360,U 12840,U
A7	REAL*8			11890,S 12130,U 12130,S 12170,U 12240,U 12300,U 12300,S 12330,U 12330,U 12340,U 12350,U 12360,U 12850,U
A8	REAL*8			11900,S 12170,S 12170,U 12240,U 12310,S 12310,U 12330,U 12330,U 12340,U 12350,U 12360,U 12860,U
A9	REAL*8			11910,S 12240,S 12240,U 12320,S 12320,U 12330,U 12330,U 12340,U 12350,U 12360,U 12870,U
BSL	REAL*8			14460,S 14550,U 14650,U
BSTL	REAL*8			14480,S 14550,U 14650,U
BSTMST	REAL*8	20		13720,S 14620,U 14630,U 14680,S
BSTRMS	REAL*8	20		13720,D 14600,U 14660,S

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
BSTRMT	REAL*8		20	13720,D 14610,U 14670,S
BSTTMT	REAL*8		20	13720,D 14620,U 14700,S
BSTU	REAL*8			14450,S 14550,U 14650,U
BSU	REAL*8			14430,S 14540,U 14650,U
BTL	REAL*8			14470,S 14550,U 14650,U
BTTMST	REAL*8		20	13720,D 14600,U 14690,S
BTU	REAL*8			14440,S 14540,U 14650,U
CARD	REAL*8		20	130,D 140,S 300,S 310,U 330,U 330,U 390,U 400,U
CCC	REAL*8		125	7090,D 8100,U 8120,U 8150,U 8820,U 17600,D 17950,S
CCCC	REAL*8		625	7100,D 8520,U 17610,D 18440,S
CC1	REAL*8			5730,D 6640,D 6780,S 7150,D 8100,U 8120,U 8150,U 8170,U 8510,U 8520,U 8820,U 13660,D 14340,U 14340,U
CC2	REAL*8			5730,D 6640,D 6790,S 7150,D 8120,U 8170,U 8540,U 8830,U 13660,D 14350,U 14350,U
CES	REAL*8			7920,S 8100,S 8100,U 8210,U 8240,U 8260,U 8290,U 16970,D 17020,S 17040,S 17060,S 17080,S 17280,U 17280,U 17280,U 17300,U 17300,U 17310,U 17320,U 17320,U 17320,U 17330,U 17340,U 17340,U 17340,U 17350,U 17360,U 17360,U 17360,U 17380,U 17380,U 17380,U 17390,U 17400,U 17400,U 17420,U 17420,U
CEST	REAL*8			7940,S 8140,U 8140,S 8210,U 8230,U 8240,U 8260,U 8280,U 8290,U
CET	REAL*8			7930,S 8120,S 8130,U 8230,U 8240,U 8280,U 8290,U
CE13	REAL*8			7950,S 8150,S 8160,U 8210,U 8240,U 8260,U 8290,U 8690,S 8820,U 8820,S 8860,U 8880,U 8890,U 8910,U
CE23	REAL*8			7960,S 8170,S 8180,U 8210,U 8230,U 8240,U 8260,U 8280,U 8290,U 8700,S 8830,U 8830,S 8860,U 8870,U 8880,U 8890,U 8900,U 8910,U
CE413	REAL*8			8340,S 8520,S 8520,S 8580,U 8600,U 8610,U 8630,U
CE423	REAL*8			8350,S 8540,U 8540,S 8580,U 8590,U 8600,U 8610,U 8620,U 8630,U
CHALS	REAL*8			930,D 14900,D 16910,D
CHECK	REAL*8		8,8	930,D 2650,S 2660,U 4310,S 4490,S 14900,D 15910,U 16910,D 17510,U
CHIS	REAL*8			13880,S 14160,U 14160,S 14370,U 14380,U
CHIST	REAL*8			13900,S 14240,U 14240,S 14390,U
CHIST1	REAL*8			14200,S 14240,U 14270,U

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
CHIS1	REAL*8			14140,S 14160,U 14250,U
CHIS2	REAL*8			14150,S 14160,U 14250,U
CHIT	REAL*8			13890,S 14190,S 14190,U 14370,U 14380,U
CHIT1	REAL*8			14170,S 14190,U 14260,U
CHIT2	REAL*8			14180,S 14190,U 14260,U
CL2R	REAL*8			7570,S 7650,U
COMENT	REAL*8		20	1130,D 1150,S 1270,S 1280,U 1490,S 1510,U
CONST	REAL*8			50,D 980,D 5710,D 6550,D 7070,D 9020,D 9830,D 10280,D 11270,D 13610,D 14930,D 16120,D 16920,D 17620,D
CONSTF	REAL*8			1160,D 1200,S 3710,U 3980,S 4000,U 4060,U 4720,U
CONSTN	REAL*8	CONSTANT		1160,D 1170,S 3710,U 4000,U 4720,U
CCNST1	REAL*8	8' * S		1160,D 1170,S 1200,U
COPH	REAL*8			7350,S 7420,U 7450,U 7470,U 7550,U 7570,U
COSINE	REAL*8		51	1030,D 2720,S 2740,S 2740,S 6610,D 7140,D 13650,D 14120,U 14130,U 16880,D
COSM	REAL*8		50	1030,D 2790,S 6610,D 7140,D 13650,D 14210,U 14220,U 16880,D
CO2R	REAL*8			7550,S 7630,U
CS	REAL*8			6120,S 6180,U 7090,D 13980,S 14040,U 14050,U 14070,U 14090,U 14100,U 14160,U 14190,U 14270,U 17600,D
CSS	REAL*8		125	7090,D 8140,U 8170,U 8830,U 17600,D 17970,S
CS4	REAL*8			7100,D 17610,D
CTHIS	REAL*8			13910,S 14250,S 14250,U 14500,U
CTHIST	REAL*8			13930,S 14270,U 14270,S 14510,U
CTHIT	REAL*8			13920,S 14260,S 14260,U 14500,U
CYCLE	REAL*8			110,D 1100,D 11300,D
C1ST	REAL*8			14410,S 14430,U 14440,U 14450,U 14460,U 14470,U 14480,U
C12	REAL*8			15740,S 15750,S 15760,U 15770,U 15780,U
C2ST	REAL*8			14420,S 14430,U 14440,U 14450,U 14460,U 14470,U 14480,U
DABS	REAL*8			6330,U

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW	USED									
DATA	REAL*8				1170,S									
DCOS	REAL*8				2790,U	6120,U	13980,U	15540,U	15540,U					
DD1	REAL*8				5730,D	6640,D	6820,S	7150,D	13660,D	14370,U	14370,U			
DD2	REAL*8				5730,D	6640,D	6830,S	7150,D	13660,D	14380,U	14380,U	14500,U	14500,U	
DELTE	REAL*8				70,D	580,U	580,U	630,U	730,U	1010,D	1320,S	1780,U	2010,U	3430,U
					4910,U	4950,U	6570,D	7190,D	9050,D	9150,U	9200,U	9860,D	13680,D	13790,U
DELTEP	REAL*8				100,D	1070,D	3280,S	3430,U	3510,U	5770,D	9060,D			
DOUBLE	REAL*8				1160,D									
DRD	REAL*8				7310,S	7330,U	7330,U	7340,U						
DSIN	REAL*8				2800,U	6110,U	13990,U	14610,U	14630,U	15520,U	15520,U	15530,U	15530,U	16610,U
					16610,U	16620,U	16620,U							
DSQRT	REAL*8				7330,U									
DTH	REAL*8		50,5,2		1080,D	3310,S	3650,S	3660,S	3780,S	3780,U	3900,U	3900,S	3920,S	4450,S
					4580,U	4780,U	4780,S	4940,U	7170,D	13710,D	14020,U	14020,U	14020,U	16150,D
					16740,S	16890,D	17270,U	17290,U	17310,U	17330,U	17350,U	17370,U	17390,U	17400,U
					17420,U									
DTHT	REAL*8				14020,S	14150,U	14180,U							
DTH1	REAL*8		5		1130,D	4420,S	4450,U							
DT2	REAL*8				50,D	980,D	2010,S	5710,D	6550,D	7070,D	9020,D	9290,U	9330,U	9350,U
					9350,U	9400,U	9650,U	9830,D	10030,U	10280,D	11270,D	13610,D	14930,D	16120,D
					16920,D	17620,D								
DUM	REAL*8		1310		1140,D	1150,S	1590,S	1610,S	1640,S	1660,S	1690,S	1730,S		
DZ	REAL*8				7320,S	7330,U	7330,U	7350,U						
EES	REAL*8				6620,D	7110,D	13600,D							
EPS	REAL*8				14300,S	14340,U	14350,U							
EPST	REAL*8				14320,S	14360,U								
EPT	REAL*8				14310,S	14340,U	14350,U							
ES	REAL*8		5		6620,D	7110,D	7820,S	8150,U	8170,U	13600,D	14040,U			
ESQ1	REAL*8				7500,S	7820,U	8210,U							
ESQ3	REAL*8				7510,S	7820,U	8240,U							



VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED													
ESQ5	REAL*8			7520,S	7820,U	8260,U											
ESQ7	REAL*8			7530,S	7820,U	8290,U											
EST	REAL*8		5	6620,D	7110,D	7830,S	8160,U	8180,U	13600,D	14060,U							
ESTQ1	REAL*8		5	7230,D	7660,S	7830,U	8210,U										
ESTQ2	REAL*8			7480,S	7830,U	8230,U											
ESTQ3	REAL*8		5	7230,D	7670,S	7830,U	8240,U										
ESTQ5	REAL*8		5	7240,D	7680,S	7830,U	8260,U										
ESTQ6	REAL*8			7490,S	7840,U	8280,U											
ESTQ7	REAL*8		5	7240,D	7690,S	7840,U	8290,U										
ESTU	REAL*8			13850,S	14060,U	14060,S	14320,U										
ESU	REAL*8			13830,S	14040,S	14040,U	14300,U										
ESUT	REAL*8			13940,S	14090,S	14090,U	14300,U										
ET	REAL*8		5	6620,D	7110,D	7800,S	8150,U	8170,U	13600,D	14050,U							
ETQ2	REAL*8		5	7240,D	7700,S	7710,U	7800,U	8230,U									
ETQ3	REAL*8			7400,S	7800,U	8240,U											
ETQ6	REAL*8		5	7240,D	7710,S	7800,U	8280,U										
ETQ7	REAL*8			7410,S	7800,U	8290,U											
ETU	REAL*8			13840,S	14050,S	14050,U	14310,U										
ETUT	REAL*8			14100,U	14100,S	14310,U											
E1	REAL*8		50	1020,D 17060,U	2680,S	3020,U	6600,D	6780,U	6820,U	7130,D	13640,D	16870,D	17020,U				
E13	REAL*8		5	6620,D 8510,U	7110,D 8520,U	7850,S 8520,U	8100,U 8540,U	8100,U 8820,U	8120,U 13600,D	8120,U 14070,U	8140,U 14170,U	8150,U 14200,U	8170,U				
E13Q1	REAL*8			7440,S	7510,U	7850,U	8210,U	8580,U	8860,U								
E13Q3	REAL*8			7450,S	7500,U	7850,U	8240,U	8600,U	8880,U								
E13Q5	REAL*8			7460,S	7530,U	7850,U	8260,U	8610,U	8890,U								
E13Q7	REAL*8			7470,S	7520,U	7850,U	8290,U	8630,U	8910,U								
E13U	REAL*8			13860,S	14070,U	14070,S	14300,U	14320,U									

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
E2	REAL*8		50	1020,D 2690,S 3020,U 6600,D 6790,U 6830,U 7130,D 13640,D 16870,D 17040,U 17080,U
E23	REAL*8		5	6620,D 7110,D 7860,S 8100,U 8100,U 8120,U 8120,U 8140,U 8160,U 8170,U 8510,U 8520,U 8540,U 8540,U 8540,U 8830,U 13600,D 14080,U 14170,U 14200,U
E23Q1	REAL*8		5	7230,D 7620,S 7670,U 7860,U 8210,U 8580,U 8860,U
E23Q2	REAL*8			7420,S 7430,U 7860,U 8230,U 8590,U 8870,U
E23Q3	REAL*8		5	7230,D 7630,S 7660,U 7860,U 8240,U 8600,U 8880,U
E23Q5	REAL*8		5	7230,D 7640,S 7690,U 7860,U 8260,U 8610,U 8890,U
E23Q6	REAL*8			7430,S 7870,U 8280,U 8620,U 8900,U
E23Q7	REAL*8		5	7230,D 7650,S 7680,U 7870,U 8290,U 8630,U 8910,U
E23U	REAL*8			13870,S 14080,S 14080,U 14310,U 14320,U
FEIGHT	REAL*8			18230,S 18390,S 18430,S 18470,U
FFIVE	REAL*8			18200,S 18360,S 18400,S 18470,U
FFOUR	REAL*8			17860,S 17930,S 17940,S 17970,U 18190,S 18300,S 18310,S 18440,U 18450,U 18460,U
FN	REAL*8			6770,S 6780,U 6790,U 6820,U 6830,U
FNU1	REAL*8		50	1020,D 2680,S 3020,U 6600,D 6770,U 7130,D 8100,U 8120,U 8150,U 8170,U 8510,U 8820,U 13640,D 14340,U 14370,U 16870,D 17020,U 17020,U 17040,U 17060,U
FNU2	REAL*8		50	1020,D 2680,S 3020,U 6600,D 6770,U 7130,D 8830,U 13640,D 14350,U 14380,U 14500,U 16870,D 17020,U 17040,U 17040,U 17060,U 17080,U 17090,U
FONE	REAL*8			17830,S 17870,S 17880,S 17950,U 17960,U 18160,S 18240,S 18250,S 18440,U 18450,U
FOR	REAL*8			8510,S 8520,U 8540,U
FORCF	REAL*8		2040	30,D 950,D 3080,S 3090,S 3280,S 3580,S 3590,S 3740,U 3740,S 3840,U 3840,S 3950,S 4190,S 4200,S 4210,S 4220,S 4700,U 4700,U 4700,U 4700,U 4740,U 4740,S 4910,U 5690,D 6580,D 7200,D 8990,D 9220,U 9240,U 9240,U 9240,U 9330,U 9350,U 9360,U 9360,U 9800,D 10010,U 10010,U 10010,U 10270,D 10720,S 10850,U 10870,D 11040,U 11050,U 11060,U 11060,U 11070,U 11070,U 11080,U 11180,U 11190,U 11210,U 13690,D 14910,D 15940,U 15940,S 16940,D 17540,U 17540,S
FRCE	REAL*8			14890,D 16140,D
FRCES	REAL*8			4320,U 14870,D

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
FSEVEN	REAL*8			18220,S 18380,S 18420,S 18470,U
FSIX	REAL*8			18210,S 18370,S 18410,S 18470,U
FTHREE	REAL*8			17850,S 17910,S 17920,S 17970,U 18180,S 18280,S 18290,S 18440,U 18450,U 18460,U
FTWO	REAL*8			17840,S 17890,S 17900,S 17950,U 17960,U 18170,S 18260,S 18270,S 18440,U 18450,U
F1	REAL*8			4160,S 4190,U 4230,U
F2	REAL*8			4160,S 4200,U 4230,U
F3	REAL*8			4160,S 4210,U 4230,U
F4	REAL*8			4160,S 4220,U 4230,U
G	REAL*8		50	1020,D 2690,S 3020,U 6600,D 6800,U 6810,U 7130,D 13640,D 16870,D
GCD	REAL*8			5730,D 6640,D 7150,D 13660,D
GDOM	REAL*8			1020,D 6600,D 7130,D 13640,D 16870,D
GG1	REAL*8			5730,D 6640,D 6800,S 7150,D 8140,U 8150,U 8170,U 8510,U 13660,D 14360,U
GG2	REAL*8			5730,D 6640,D 6810,S 7150,D 13660,D 14390,U 14510,U
HARM	REAL*8			90,D 1060,D 5760,D 7160,D 13670,D 14950,D 16160,D 16900,D 17640,D
HOUBON	REAL*8			5870,U 9770,D
HOUBQ1	REAL*8			5860,U 8960,D

VARIABLE TYPE INITIAL VALUE DIMENSION WHERE/HOW USED

VARIABLE TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
I	INTEGER		
	510,0	520,U	1250,S 1340,U 1340,S 1370,S 1370,U 1590,U 1590,S 1610,S
	1610,U	1640,U	1640,S 1660,S 1660,U 1690,U 1690,S 1730,S 1730,U 1740,U
	1740,S	1800,S	1800,U 1810,U 1810,S 1930,S 1940,U 1940,U 2020,S 2030,U
	2040,U	2140,U	2180,U 2550,S 2560,U 2590,U 2650,S 2650,U 2650,S 2650,U
	2660,S	2660,U	2660,S 2660,U 2680,S 2680,U 2680,S 2680,U 2680,S 2680,U
	2690,S	2690,U	2690,S 2690,U 2690,S 2690,U 2690,S 2690,U 2700,S 2710,U
	2720,U	2720,U	2720,U 2720,U 2740,U 2740,U 2740,U 2740,U 2740,U 2740,U
	2740,U	2750,U	2770,S 2770,U 2770,S 2770,U 2770,S 2770,U 2780,S 2790,U
	2800,U	3020,U	3020,U 3020,U 3020,U 3020,U 3020,U 3020,U 3020,U 3020,U
	3030,U	3030,U	3030,S 3030,U 3030,S 3080,S 3080,U 3080,S 3090,U 3130,U
	3130,S	3140,S	3140,U 3170,U 3170,S 3190,S 3190,U 3270,U 3270,S 3280,S
	3280,U	3290,U	3290,S 3320,U 3320,S 3320,U 3320,S 3320,U 3320,S 3320,U
	3380,U	3390,U	3390,U 3400,U 3410,U 3420,U 3420,U 3430,U 3430,U 3430,U
	3570,S	3580,U	3590,U 3610,S 3630,U 3640,U 3650,U 3660,U 3730,U 3740,S
	3740,S	3750,S	3770,U 3770,U 3780,U 3780,U 3830,U 3840,U 3840,U 3840,U
	3870,S	3890,U	3890,U 3900,U 3900,U 3910,U 3920,U 4310,U 4310,S 4310,U
	4310,S	4490,U	4490,S 4490,U 4490,S 4680,S 4690,U 4700,U 4730,S 4740,U
	4740,U	4750,S	4770,U 4770,U 4780,U 4780,U 4780,U 4900,U 4910,S 4910,U
	4920,U	4980,U	4980,S 4990,U 4990,U 4990,U 4990,U 5000,U 5000,U 5000,U
	5000,U	5000,S	5000,S 5960,S 5970,U 5980,U 5980,U 6060,S 6070,U 6250,S
	6270,U	6730,S	6740,U 7980,S 7990,U 8010,S 8020,U 8100,U 8100,U 8120,U
	8120,U	8140,U	8150,U 8160,U 8170,U 8180,U 8370,S 8380,U 8510,U 8520,U
	8540,U	8720,S	8730,U 8820,U 8830,U 9180,S 9190,U 9220,U 9220,U 9240,U
	9240,U	9300,S	9310,U 9330,U 9330,U 9330,U 9350,U 9350,U 9350,U 9380,S
	9390,U	9470,S	9480,U 9490,U 9500,U 9500,U 9520,S 9530,U 9630,S 9640,U
	9650,U	9670,S	9680,U 9920,S 9930,U 9990,S 10000,U 10030,U 10030,U 10120,S
	10130,U	10190,S	10200,U 10210,U 10420,S 10440,U 10440,U 10440,U 10470,S 10480,U
	10480,U	10480,U	10540,S 10550,U 10570,U 10570,U 10570,U 10570,U 10610,U 10620,U
	10620,U	10620,U	10620,U 10690,S 10700,U 10700,U 10700,U 10700,U 10800,S 10810,U
	10940,S	10950,U	11020,S 11040,U 11040,U 11110,S 11120,U 11390,S 11400,U 11410,U
	11420,S	11430,U	11430,U 11620,S 11650,U 11660,U 11670,U 11680,U 11690,U 11700,U
	11720,U	11730,U	11740,U 11750,U 11760,U 11770,U 11780,U 11790,U 11800,U 11810,U
	11820,U	11830,U	11840,U 11850,U 11860,U 11870,U 11880,U 11890,U 11900,U 11910,U
	11920,U	11930,U	11940,U 11950,U 11960,U 11970,U 11980,U 11990,U 12000,U 12010,U
	12020,U	12030,U	12040,U 12050,U 12060,U 12070,U 12600,U 12610,U 12620,U 12630,U
	12640,U	12650,U	12660,U 12690,U 12700,U 12710,U 12720,U 12730,U 12740,U 12750,U
	12760,U	12770,U	12780,U 12790,U 12800,U 12810,U 12820,U 12830,U 12840,U 12850,U
	12860,U	12870,U	12880,U 12890,U 12900,U 12910,U 12920,U 12930,U 12940,U 12950,U
	12960,U	12970,U	12980,U 12990,U 13000,U 13010,U 13020,U 13030,U 13500,S 13510,U
	13510,U	13530,S	13540,U 13550,U 13820,S 13980,U 13990,U 14520,U 14600,U 14600,U
	14610,U	14620,U	14620,U 14630,U 14660,U 14670,U 14680,U 14690,U 14700,U 15080,S
	15080,U	15080,U	15080,U 15090,S 15090,U 15090,U 15090,U 15090,U 15170,U 15170,U
	15170,U	15170,U	15170,S 15270,U 15270,U 15270,U 15270,U 15280,U 15280,U 15310,U
	15310,U	15310,U	15320,U 15320,S 15320,U 15400,S 15410,U 15430,U 15450,U 15460,U
	15480,U	15480,U	15480,U 15490,U 15490,U 15490,U 15520,U 15530,U 15540,U 15880,S
	15890,U	15910,U	15910,U 15910,U 15920,S 15930,U 15940,U 16300,U 16300,S 16300,U
	16300,U	16390,U	16390,U 16400,S 16400,U 16400,U 16520,S 16530,U 16530,U 16530,U
	16540,U	16540,U	16540,U 16580,S 16590,U 16600,U 16610,U 16620,U 17480,S 17490,U
	17510,U	17510,U	17510,U 17520,S 17530,U 17540,U
IABS	INTEGER		8070,U 8420,U 8460,U 8780,U 17780,U 17820,U 18080,U 18120,U 18330,U 18350,U

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VARIABLE	TYPE	INITIAL VALUF	DIMENSION	WHERE/HOW USED
IB	INTEGER			3940,S 3950,S 3960,U 4000,U 4180,U 4320,P 4390,U 4510,P 4520,P 4650,U 4720,U 4820,U 4820,S 4830,U 14870,U 15930,U 16100,U 16720,U 16850,U 17230,U 17530,U
IBP1	INTEGER			3300,U 3300,S 3310,S 3310,U 4390,S 4450,U 4460,U 4580,U 4580,U 4930,S 4930,U 4940,S 4940,U 16720,S 16730,U 16740,U 17230,S 17270,U 17270,U 17280,U 17290,U 17300,U 17310,U 17320,U 17330,U 17340,U 17350,U 17360,U 17370,U 17380,U 17390,U 17400,U 17410,U 17420,U 17430,U
IDCOE	INTEGER			60,D 990,D 3980,S 5720,D 6560,D 7080,D 9030,D 9840,D 10290,D 11280,D 13620,D 14940,D 15050,U 15390,U 16130,D 16930,D 17630,D
IDELF	INTEGER			60,D 990,D 3980,S 4280,U 5720,D 6560,D 7080,D 9030,D 9840,D 10290,D 11280,D 13620,D 14940,D 16130,D 16930,D 17630,D
IDP	INTEGER			15330,S 15340,U 15340,U 16420,S 16430,U 16430,U
IELM	INTEGER			2880,S 2890,U 2900,U 2930,S 2940,U 2950,U 3300,U 3300,S 3310,S 3310,U 4300,S 4320,P 4430,S 4450,U 4460,S 4480,S 4510,P 4520,P 4570,S 4580,U 4530,U 4580,U 4930,U 4930,S 4940,U 4940,U 10400,S 10410,U 10540,U 10570,U 10590,U 10600,U 10620,U 10640,U 10660,U 10670,U 10670,S 10700,U 10700,U 10700,U 15040,U 15040,U 15060,U 15130,U 15930,U 16100,U 16220,U 16240,U 16240,U 16250,U 16730,U 16740,U 16850,U 17020,U 17020,U 17020,U 17020,U 17020,U 17030,U 17030,U 17040,U 17040,U 17040,U 17040,U 17040,U 17050,U 17050,U 17060,U 17060,U 17060,U 17060,U 17070,U 17070,U 17070,U 17070,U 17080,U 17080,U 17080,U 17080,U 17090,U 17090,U 17090,U 17130,U 17140,U 17150,U 17150,U 17150,U 17150,U 17150,U 17160,U 17160,U 17160,U 17160,U 17190,U 17190,U 17190,U 17190,U 17190,U 17210,U 17210,U 17210,U 17210,U 17210,U 17270,U 17280,U 17290,U 17300,U 17310,U 17320,U 17330,U 17340,U 17350,U 17360,U 17370,U 17380,U 17390,U 17400,U 17400,U 17420,U 17420,U 17530,U
IELM1	INTEGER			2870,S 2880,U 4420,S 4430,U 15080,S 15090,U 15170,S 15270,U 16300,S 16390,U
IELM2	INTEGER			2870,S 2880,U 2910,U 4420,S 4430,U 4470,U 15030,S 15040,U 15080,S 15090,U 15170,S 15270,U 16220,S 16240,U 16300,S 16390,U
IEQ	INTEGER			10130,S 10140,U
IFLAG	INTEGER			2320,S 2330,U 2340,U 2350,U 2360,U 2460,S 2470,U 2480,U 2490,U 2500,U 9130,S 9460,U 9590,S
IFO	INTEGER			8320,S 8490,S 8490,U 8520,U 8520,U 8540,U 8540,U 18000,S 18150,S 18150,U 18440,U 18450,U 18460,U 18470,U

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED		
IH	INTEGER			2270,S 2280,U 2410,S 2420,U 3040,S 3060,U 3300,U 3300,S 3310,S 3310,S 3310,S 3310,S		
				4080,S 4090,U 4140,U 4420,U 4420,S 4420,U 4440,S 4440,S 4450,U 4450,U 4460,U 4460,U		
				4460,U 4550,S 4560,U 4580,U 4580,U 4640,S 4650,U 4660,U 4930,S 4930,S 4930,S 4930,S		
				4940,U 4940,S 5830,S 5840,U 5850,U 5860,U 5870,P 5880,P 6000,U 6000,U 6000,U 6000,U		
				7600,U 7620,U 7630,U 7640,U 7650,U 7660,U 7660,U 7670,U 7670,U 7670,U 7670,U 7680,U		
				7680,U 7690,U 7690,U 7700,U 7710,U 7710,U 7730,U 7800,U 7800,U 7800,U 7800,U 7800,U		
				7820,U 7830,U 7830,U 7830,U 7830,U 7840,U 7850,U 7860,U 7860,U 7860,U 7860,U 7860,U		
				7860,U 7870,U 8960,U 9170,P 9280,P 9450,P 9770,U 9960,P 10180,P 10850,U 10850,U 10850,U		
				10980,U 10990,U 11240,U 13960,S 13970,U 14000,U 14010,U 14010,U 14010,U 14010,U 14010,U 14020,U		
				14020,U 14120,U 14040,U 14050,U 14060,U 14070,U 14080,U 14170,U 14170,U 14170,U 14200,U 14200,U		
				14200,U 15360,S 15370,U 15760,U 15770,U 15780,U 15930,U 16450,S 16460,U 16730,U 16730,U 16730,U		
				16740,U 17240,S 17250,U 17270,U 17270,U 17280,U 17290,U 17300,U 17310,U 17320,U 17320,U 17320,U		
				17330,U 17340,U 17350,U 17360,U 17370,U 17380,U 17390,U 17400,U 17410,U 17420,U 17420,U 17420,U		
				17430,U 17530,U		
		IHARM	INTEGER		5	90,D 1060,D 1370,U 1740,U 1800,U 2590,U 3060,U 3160,U 3180,U 3270,S 3270,S 3270,S
						4140,U 4560,U 4660,U 4900,U 5760,D 5840,U 6100,U 7160,D 7600,U 7970,U 7970,U 7970,U
						8020,U 8040,U 8360,U 8380,U 8400,U 8440,U 8710,U 8730,U 8750,U 13670,D 13670,D 13670,D
						13970,U 14950,D 15370,U 16160,D 16460,U 16900,D 17250,U 17640,D 17720,U 17740,U 17740,U 17740,U
						17760,U 18020,U 18040,U 18060,U 18100,U
		IH1	INTEGER		4090,S 4180,U	
II	INTEGER			2570,S 2580,U 2590,U 2640,S 6950,S 6960,U 6990,U 7000,U 8020,S 8050,U 8050,U 8050,U		
				8060,U 8380,S 8410,U 8420,U 8730,S 8760,U 8770,U 17740,S 17770,U 17780,U 17780,U		
IID	INTEGER			10850,U 10940,U 10960,U 10990,U		
				17900,U 18080,S 18130,U 18130,U 18280,U 18290,U 18300,U 18310,U		
IMJ	INTEGER			8060,S 8070,U 8420,S 8470,U 8470,U 8770,S 8780,U 17780,S 17790,U 17890,U 17890,U		
				17900,U 18080,S 18130,U 18130,U 18280,U 18290,U 18300,U 18310,U		
IML	INTEGER			18350,S 18360,U 18370,U 18400,U 18410,U		
IMM	INTEGER			17820,S 17930,U 17940,U		
IN	INTEGER			4170,S 4180,U 4230,U		
INCRST	INTEGER			1320,S 1780,U 3350,U		
INODE	INTEGER			2310,S 2320,U 2450,S 2460,U		
INPUT	INTEGER			470,U 670,D 910,D 6400,U		
IN1	INTEGER			2300,S 2310,U 2440,S 2450,U 4160,S 4170,U		
IN2	INTEGER			2300,S 2310,U 2370,U 2440,S 2450,U 2510,U 4160,S 4170,U 4250,U		
IPJ	INTEGER			8050,S 8070,U 8410,S 8470,U 8470,U 8760,S 8780,U 17770,S 17790,U 17870,U 17870,U		
				17880,U 18070,S 18130,U 18130,U 18240,U 18250,U 18260,U 18270,U		
IPL	INTEGER			18340,S 18380,U 18390,U 18420,U 18430,U		
IPM	INTEGER			17810,S 17910,U 17920,U		

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
IPRINT	INTEGER			80,D 1050,D 1320,S 1780,U 5750,D 5890,U 5890,U 6660,D 11290,D
IQ	INTEGER			2560,S 2590,U 2590,U 2590,U 2590,U 2600,U 2600,U 2600,U 2600,U
ION	INTEGER			2250,S 2260,U
ION1	INTEGER			2250,S 2400,U
IRSTR	INTEGER			100,D 500,U 610,U 1070,D 1320,S 1370,U 1390,U 1780,U 2060,U 3220,U 5770,D 9060,D 9210,U 9320,U
IT	INTEGER			5890,S 5910,U 6050,S 6110,U 6120,U 6230,U
ITAM	INTEGER			110,D 620,S 630,U 650,U 690,P 720,S 740,P 1100,D 4910,U 5020,U 5660,U 5820,U 5820,P 5870,U 5880,U 5890,U 5890,U 5890,U 5900,U 5910,U 5940,U 6220,U 6350,U 6390,U 6390,U 6530,U 6920,U 6920,U 6920,U 6920,U 6930,P 11300,D 11440,U 13520,U 13580,U 13780,U
ITAM1	INTEGER			13780,S 13810,U
ITCOE	INTEGER			1090,D 3980,S 4400,U 4500,U 7180,D
ITELF	INTEGER			1090,D 1320,S 1790,U 2820,U 4340,U 7180,D 8650,U
ITH	INTEGER			7890,S 8080,U 8080,S 8100,U 8100,U 8120,U 8120,U 8140,U 8150,U 8150,U 8170,U 8660,S 8800,S 8800,U 8820,U 8830,U 17700,S 17800,S 17800,U 17950,U 17960,U 17970,U
ITP	INTEGER			100,D 550,S 580,U 610,U 630,U 1070,D 3280,S 3350,U 3510,U 5770,D 9060,D
IT1	INTEGER			6390,S 6400,U
IX	INTEGER			2590,S 2590,U 2590,U 2590,U 2600,U 2600,U 2600,U 2600,U 2600,U
I1	INTEGER			6130,S 6140,U 6760,S 6770,U 6770,U 6780,U 6780,U 6790,U 6790,U 6800,U 6800,U 6810,U 6810,U 6820,U 6820,U 6830,U 6830,U 6870,P 6930,P 6960,U 7050,U 7290,U 7300,U 7730,U 8100,U 8120,U 8150,U 8170,U 8510,U 8790,U 8790,U 8790,U 8820,U 8820,U 8820,U 8830,U 8830,U 8830,U 13580,U 13810,U 14050,U 14010,U 14010,U 14010,U 14020,U 14020,U 14020,U 14090,U 14100,U 14120,U 14120,U 14130,U 14130,U 14140,U 14150,U 14170,U 14170,U 14180,U 14200,U 14200,U 14210,U 14210,U 14210,U 14210,U 14220,U 14220,U 14220,U 14220,U 14220,U 14340,U 14350,U 14370,U 14380,U 14410,U 14420,U 14500,U 14530,U 14540,U 14570,U 14570,U 14580,U 14580,U 14590,U 14590,U 14600,U 14600,U 14620,U 14620,U 14640,U

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
J	INTEGER			1270,S 1270,U 1280,U 1280,S 1490,S 1490,U 1510,U 1510,S 1600,U 1610,U 1680,S 1690,U 1700,S 1720,U 2650,U 2650,S 2660,S 2660,U 3620,S 3630,U 3660,U 3650,U 3660,U 3760,S 3770,U 3770,U 3780,U 3780,U 3880,S 3890,U 3890,U 3900,U 3900,U 3910,U 3920,U 4310,S 4310,U 4490,U 4490,S 4760,S 4770,U 4770,U 4780,U 4780,U 5980,S 5980,U 6160,S 6170,U 6180,U 6180,U 6180,U 6200,U 6200,U 6200,U 6200,U 6270,U 6270,U 8030,S 8040,U 8100,U 8100,U 8120,U 8120,U 8140,U 8150,U 8150,U 8160,U 8170,U 8170,U 8170,U 8390,S 8400,U 8520,U 8520,U 8540,U 8540,U 8740,S 8750,U 8790,U 8790,U 8790,U 10430,S 10440,U 10460,S 10480,U 10560,S 10570,U 10590,S 10620,U 10640,U 10680,U 10680,S 10700,U 11010,S 11020,U 11040,U 11140,U 11150,U 11180,U 11190,U 11190,U 11210,U 11210,U 11210,U 11630,S 12270,U 12270,U 12270,U 12270,U 12270,U 12270,U 12360,U 12360,U 12360,U 12360,U 12360,U 12360,U 12360,U 12460,U 12460,U 12460,U 12460,U 12460,U 12460,U 12460,U 12470,U 12570,U 12570,U 12570,U 12570,U 12570,U 12570,U 12570,U 12580,U 12580,U 15900,S 15910,U 15910,U 15930,S 15940,U 15940,U 17460,S 17470,U 17470,U 17500,S 17510,U 17510,U 17530,U 17540,U 17540,U
JH	INTEGER			3050,S 3060,U 3120,U 3160,U 3180,U 6080,S 6090,S 6100,U
JJ	INTEGER			6970,S 6980,U 6990,U 7000,U 8040,S 8050,U 8060,U 8400,S 8410,U 8420,U 8750,S 8760,U 8770,U 17760,S 17770,U 17780,U 17910,U 17920,U 17920,U 17930,U 17940,U 17940,U 18060,S 18070,U 18080,U 18320,U 18330,U
JM1	INTEGER			11150,S 11160,U
JUNK	INTEGER	20		1130,D 1450,S 1530,S 1740,S 3270,S 4900,U
J1	INTEGER			7290,S 7310,U 7320,U 7360,U
J11	INTEGER			7300,U 7310,U 7320,U 7360,U
K	INTEGER			1480,S 1580,S 1620,S 1630,U 1720,S 4180,S 4190,U 4200,U 4210,U 4220,U 4690,S 4730,U 4700,U 4700,U 4700,U 5970,S 5980,U 6140,S 6150,U 6180,U 6180,U 6200,U 6200,U 6260,U 6270,U 7600,S 7610,U 7970,S 8070,U 8070,U 8430,S 8440,U 8510,U 8520,U 8540,U 8710,S 8780,U 8780,U 10440,S 10450,U 10480,S 10490,U 10500,S 10510,U 10570,S 10580,U 10620,S 10630,U 10640,S 10650,U 10700,S 10710,U 10810,S 10820,U 11610,S 11620,U 11630,U 13090,S 13100,U 13110,U 13280,S 13290,U 13300,U 14000,S 14120,U 14120,U 14130,U 14130,U 14140,U 14140,U 14210,U 14210,U 14220,U 14220,U
KA	INTEGER			6960,S 6980,U
KEY	INTEGER			910,U 1180,U 3710,U 3950,U 11450,S 11640,U 12590,U 12660,S 15210,S 15220,U 15230,U 15240,U 15250,U 16340,S 16350,U 16360,U 16370,U
KEYRS	INTEGER			1380,S 1390,S 1500,U 1540,U 1750,S
KK	INTEGER			4650,S 4690,U 6980,S 6990,U 6990,U 7000,U 7730,S 7740,U 7750,U 7760,U 7770,U 7780,U 7790,U 8440,S 8450,U 8460,U 18100,S 18110,U 18120,U 18320,U 18330,U
KKP2	INTEGER			580,S 590,U 620,U 650,U 720,U
KK1	INTEGER			7740,U 7820,U 7830,U 7850,U 7860,U



VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
KK2	INTEGER			7750,S 7800,U 7830,U 7860,U
KK3	INTEGER			7760,S 7800,U 7820,U 7830,U 7850,U 7860,U
KK5	INTEGER			7770,S 7820,U 7840,U 7850,U 7870,U
KK6	INTEGER			7780,S 7800,U 7840,U 7870,U
KK7	INTEGER			7790,S 7800,U 7820,U 7840,U 7850,U 7870,U
KMJ	INTEGER			18330,S 18370,U 18390,U 18410,U 18410,U 18430,U 18430,U
KML	INTEGER			8460,S 8470,U 8470,U 18120,S 18130,U 18130,U 18260,U 18270,U 18270,U 18300,U 18310,U 18310,U
KPJ	INTEGER			18320,S 18360,U 18380,U 18400,U 18400,U 18420,U 18420,U
KPL	INTEGER			8450,S 8470,U 8470,U 18110,S 18130,U 18130,U 18240,U 18250,U 18250,U 18280,U 18290,U 18290,U
KY	INTEGER			5840,S 5870,P 5880,P 5950,U 8960,U 9430,P 9740,P 9770,U 10100,U 10250,U 10360,U 10780,U
KYP	INTEGER			4660,S 4670,U 15370,S 15380,U 15470,U 15560,U 15750,U 16460,S 16470,U 16510,U 16660,U 17250,S 17260,U 17440,U
L	INTEGER			8330,S 8360,U 8580,U 8580,U 8580,U 8590,U 8590,U 8600,U 8600,U 8600,U 8610,U 8610,U 8610,U 8620,U 8620,U 8630,U 8630,U 8630,U 10380,S 10390,U 11160,S 11180,U 11180,U 11190,U 13110,U 13120,U 13120,U 13120,U 13120,U 13130,U 13130,S 13130,U 13130,U 13140,U 13150,U 13150,U 13150,U 13160,U 13160,U 13160,U 13170,U 13170,U 13170,U 13180,U 13180,U 13180,U 13180,U 13290,S 13310,U 13320,U 13320,U 13320,U 13320,U 13330,U 13340,U 13340,U 13340,U 13340,U 13350,U 13350,U 13360,U 13360,U 13360,U 13360,U 13370,U 13370,U 13370,U 13380,U 13380,U 13380,U 13380,U 13390,U 13390,U 13390,U 13390,U 13410,U 13410,U 13410,U 13410,U 13420,U 13430,U 13430,U 13430,U 13430,U 13440,U 13440,U 13450,U 13450,U 13450,U 13450,U 13460,U 13460,U 13460,U 13470,U 13470,U 13470,U 13470,U 13480,U 13480,U 13480,U 13480,U
LARGE	INTEGER			490,S 690,P 700,U 5660,U 6340,S
LE	INTEGER			6330,U 15040,U 16240,U
LK	INTEGER	204		970,D 2180,S 2190,S 2200,S 3280,S 3330,S 3340,S 4910,U 9010,D 9820,D 10130,U 10300,D 10390,U
LL	INTEGER			80,D 600,S 610,S 620,U 1050,D 5750,D 5820,U 5870,U 5880,U 6660,D 8360,S 8450,U 8460,U 11290,D 11440,U 13520,U 18020,S 18110,U 18120,U 18340,U 18350,U

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HTW USED
M	INTEGER			7910,S 7970,U 7990,U 8210,U 8210,U 8210,U 8220,U 8230,U 8230,U 8230,U 8230,U 8240,U 8240,U 8240,U 8250,U 8260,U 8260,U 8260,U 8270,U 8280,U 8280,U 8280,U 8280,U 8290,U 8290,U 8290,U 8300,U 8680,U 8710,U 8860,U 8860,U 8860,U 8860,U 8870,U 8870,U 8880,U 8880,U 8880,U 8890,U 8890,U 8890,U 8890,U 8900,U 8900,U 8910,U 8910,U 8910,U 10260,D 13230,S 13240,U 13240,U 13240,U 13250,U 13250,U 13250,U 13260,U 13260,U 13260,U 13260,U 13270,U 13270,S 13270,U 13270,U 13270,U
MATMUT	INTEGER			9170,U 9280,U 9960,U 10850,D
MPRINT	INTEGER			290,S 340,U 370,S 380,S 380,U
N	INTEGER			50,D 980,D 2280,S 2320,U 2420,S 2460,U 5710,D 5860,S 5970,U 6090,S 6150,U 6330,U 6550,D 7070,D 9020,D 9190,U 9310,U 9480,U 9640,U 9830,D 9930,U 10090,U 10200,U 10550,S 10560,U 10600,S 10610,U 10640,U 10640,U 10640,U 10640,U 13100,U 13120,U 13120,U 13120,U 13120,U 13120,U 13120,U 13120,U 13120,U 13120,U 13130,U 13130,U 13130,U 13130,U 13140,U 13150,U 13150,U 13150,U 13150,U 13150,U 13150,U 13150,U 13160,U 13160,U 13170,U 13170,U 13170,U 13170,U 13170,U 13170,U 13170,U 13170,U 13180,U 13180,U 13220,S 13240,U 13250,U 13250,U 13260,U 13260,U 13260,U 13270,U 13270,U 13270,U 13270,U 13300,S 13320,U 13320,U 13320,U 13320,U 13320,U 13320,U 13340,U 13340,U 13340,U 13340,U 13340,U 13340,U 13350,U 13360,U 13360,U 13360,U 13360,U 13360,U 13360,U 13370,U 13370,U 13370,U 13380,U 13380,U 13380,U 13380,U 13380,U 13380,U 13380,U 13390,U 13390,U 13390,U 13410,U 13410,U 13410,U 13410,U 13410,U 13410,U 13410,U 13430,U 13430,U 13430,U 13430,U 13430,U 13430,U 13440,U 13440,U 13450,U 13450,U 13450,U 13450,U 13450,U 13450,U 13460,U 13460,U 13470,U 13470,U 13470,U 13470,U 13470,U 13470,U 13470,U 13480,U 13480,U 13480,U 13610,D 14930,D 16120,D 16920,D
NA	INTEGER			11000,S 11030,S 11030,S 11040,U
NCARD	INTEGER			270,S 310,U 320,U 320,S 410,U
NCARDS	INTEGER			1230,S 1240,U 1250,U 1450,S 1460,U 1480,U
NCASE	INTEGER			220,S 280,U -280,S 350,U 420,S 420,U 430,U 460,S 480,S 480,U 750,U
NCASES	INTEGER			200,S 210,U 430,U 750,U
NCF	INTEGER			3980,S 4070,U
NCF1	INTEGER			4110,S 4120,U
NCLCST	INTEGER			1040,D 1330,S 1790,U 5740,D 6650,D 6930,U 13630,D
NCLOSE	INTEGER			970,D 1320,S 1790,U 9010,D 9820,D 10100,U 10300,D 10360,U
ND	INTEGER			120,D 200,S 200,U 330,U 440,U 1110,D 1230,U 1270,U 1320,U 1340,U 1370,U 1790,U 2110,U 2160,U 2250,U 2300,U 2440,U 2870,U 3980,U 4110,U 4160,U 4420,U 14970,D 15080,U 15170,U 16170,D 16300,U
NDIRCT	INTEGER			2160,S 2170,U 2160,U
NDP	INTEGER			15170,U 15170,S 15180,U 15190,U 15280,U 15290,U 15300,U 15310,U 15420,U 15440,U 16300,S 16300,U 16310,U 16320,U 16330,U 16400,U 16410,U 16500,U 16520,U 16580,U

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
NDPP2	INTEGER			15300,S 15320,U
NDP1	INTEGER			15420,S 15430,U
NDP2	INTEGER			15180,S 15210,U 16310,S 16340,U
ND2	INTEGER			15290,S 15320,U 15330,U 16410,S 16420,U
NELEMS	INTEGER			50,D 980,D 1530,S 1580,U 1600,U 1620,U 1630,U 1680,U 1710,U 1790,U 1960,U 1990,U 2640,U 2680,U 2680,U 2690,U 2690,U 2690,U 2700,U 2700,U 2710,U 2730,U 2910,U 2930,U 3030,U 3300,U 3310,U 3610,U 3750,U 3870,U 4300,U 4470,U 4480,U 4570,U 4750,U 4930,U 4940,U 5710,D 6550,D 6760,U 7070,D 9020,D 9830,D 10280,U 10660,D 11270,D 11610,U 13090,U 13220,U 13230,U 13280,U 13290,U 13300,U 13610,D 14930,D 16120,U 16920,D 17140,U 17620,D
NEQ	INTEGER			50,D 980,D 1970,S 1980,U 2280,U 2420,U 2560,U 4180,U 4650,U 5710,D 5860,U 6060,U 6090,U 6550,D 6960,U 7070,D 7730,U 9020,D 9170,P 9180,U 9280,P 9300,U 9470,U 9630,U 9830,D 9920,U 9960,P 9990,U 10190,U 10390,U 10400,U 10420,U 10430,U 10460,U 10470,U 10500,U 10500,U 10500,U 10500,U 10540,U 10590,U 10600,U 10720,U 11270,D 11420,U 13500,U 13610,D 14000,U 14930,D 15930,U 16120,D 16920,D 17530,U 17620,D
NEQT	INTEGER			50,D 510,U 980,D 1980,S 2020,U 3260,U 3320,U 3320,U 3320,U 3370,U 3570,U 3590,U 3730,U 3740,U 3930,U 3840,U 3850,U 4180,U 4650,U 4730,U 4740,U 4890,U 4980,U 5000,U 5000,U 5000,U 5710,D 6550,D 6730,U 7070,D 9020,D 9240,U 9360,U 9830,D 10010,U 10280,D 11270,D 13610,D 14930,D 15930,U 16120,D 16920,D 17530,U 17620,D
NF	INTEGER			10990,S 11040,U 11050,U 11060,U 11060,U 11070,U 11070,U 11080,U 11130,U
NFF	INTEGER			11130,S 11180,U 11190,U 11210,U
NH	INTEGER			50,D 980,D 1370,U 1370,S 1740,U 1740,S 1800,U 1800,U 1980,U 2000,U 2270,U 2410,U 2550,U 3050,U 3270,U 3270,S 3300,U 3310,U 3620,U 3760,U 3880,U 4080,U 4420,U 4440,U 4550,U 4640,U 4760,U 4900,U 4910,U 4930,U 4940,U 5710,D 5830,U 6000,U 6010,U 6080,U 6550,D 6950,U 7070,D 7590,U 7910,U 8010,U 8030,U 8330,U 8370,U 8390,U 8430,U 8680,U 8720,U 8740,U 9020,D 9830,D 10280,D 11270,U 13610,D 13960,U 14930,D 15080,U 15090,U 15360,U 16120,D 16450,U 16920,D 17240,U 17620,D 17710,U 17730,U 17750,U 18010,U 18030,U 18050,U 18090,U
NHNS	INTEGER			50,D 980,D 2000,S 5710,D 6550,D 7070,D 9020,D 9830,D 10280,D 11270,D 13610,D 14930,D 15120,D 16920,D 17620,D
NHP	INTEGER			90,D 1060,D 1530,S 1700,U 3040,U 5760,D 7160,D 13670,D 14950,D 16160,D 16900,D 17640,D
NI	INTEGER			11120,S 11130,U 11180,U 11180,U 11190,U 11190,U 11210,U 11210,U
NIX	INTEGER			10280,D
NK	INTEGER			10980,S 11000,U 11050,U 11060,U 11060,U 11070,U 11070,U 11070,U 11090,U
NKK	INTEGER			11090,S 11170,U 11170,S 11180,U 11190,U 11200,S 11200,S 11210,U

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
NLTERM	INTEGER			740,U 3360,U 5820,U 6530,D
NLTRMS	INTEGER			6630,D 7120,D
NM1	INTEGER			11100,S 11110,U
NN	INTEGER			50,D 980,D 3120,S 3130,U 3140,U 3170,U 3190,U 5710,D 5850,S 6550,D 7070,D 9020,D 9390,U 9530,U 9680,U 9830,D 10280,D 10440,U 10480,U 10500,U 10570,U 10620,U 10640,U 10700,U 10960,S 10970,U 11100,U 11270,D 11400,U 13540,U 13610,D 14930,D 16120,D 16920,D 17620,D
NNODES	INTEGER			50,D 980,D 1960,S 1970,U 2370,U 2510,U 2570,U 2770,U 2770,U 4250,U 4680,U 5710,D 5960,U 6130,U 6250,U 6550,D 7070,D 9020,D 9830,D 10280,U 10800,D 11270,D 13610,D 14930,D 16120,D 16920,D 17620,D
NNPI	INTEGER			9390,S 9400,U 9400,U 9400,U 9530,S 9540,U 9550,U 9550,U 9560,U 9680,S 9690,U 9700,U 9700,U 9700,U 9710,U 11400,S 11410,U 13540,S 13550,S
NODRE	INTEGER			10090,S 10100,S 10100,U 10110,U 10120,U 10350,S 10360,S 10360,U 10370,U 10380,U
NODRES	INTEGER			970,D 2110,S 2120,U 2130,U 2140,U 2190,U 2200,U 3280,U 3280,S 3330,U 3340,U 4910,U 4910,U 9010,D 9820,D 10090,U 10300,U 10350,D
NOIT	INTEGER			80,D 590,S 1050,D 1350,S 5750,D 5900,U 6660,D 6930,U 11290,D
NP	INTEGER			2160,S 2170,U 2180,U
NPI	INTEGER			9190,S 9200,U 9200,U 9200,U 9220,U 9240,U 9240,U 9240,U 9250,U 9310,S 9330,U 9350,U 9350,U 9360,U 9360,U 9480,S 9490,U 9640,S 9650,U 9650,U 9650,U 9930,S 9940,U 9940,U 9940,U 9940,U 10000,S 10010,U 10010,U 10010,U 10010,U 10010,U 10040,U 10040,U 10050,U 10050,U 10060,U 10060,U 10200,S 10210,U
NPK	INTEGER			6150,S 6180,U 6200,U
NPRNIT	INTEGER			100,D 1070,D 1330,S 1780,U 5770,D 6390,U 6390,U 9060,D
NPRNMS	INTEGER			1330,S 1800,U 3150,U
NPRNT	INTEGER			100,D 1070,D 1330,S 1780,U 5770,D 6380,U 6400,U 9060,D
NPRNTF	INTEGER			60,D 990,D 1330,S 1790,U 4010,U 4600,U 5720,D 6560,D 7080,D 9030,D 9840,D 10290,D 11280,D 13620,D 14940,D 16130,D 16930,D 17630,D
NPRNTH	INTEGER			1090,D 1330,S 1790,U 4540,U 7180,D
NPRNTL	INTEGER			60,D 990,D 1330,S 1790,U 4010,U 4130,U 4230,U 5720,D 6560,D 7080,D 9030,D 9840,D 10290,D 11280,D 13620,D 14940,D 15060,U 15090,U 15130,U 15270,U 15310,U 16130,D 16250,U 16390,U 16930,D 17630,D
NPRNTQ	INTEGER			50,D 980,D 1320,S 1780,U 5710,D 5920,U 6550,D 7070,D 9020,D 9830,D 10280,D 11270,D 13610,D 14930,D 16120,D 16920,D 17620,D

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
NPESTR	INTEGER			9430,U 9740,U 10250,D
NS	INTEGER			120,D 200,S 1110,D 1790,U 2630,U 2660,U 4290,U 4310,U 4380,U 4490,U 14970,D 16170,D
NSIZE	INTEGER			50,D 980,D 1710,S 1730,U 1990,S 2000,U 3080,U 3090,U 3120,U 3130,U 3140,U 3170,U 3190,U 5710,D 5850,U 6550,D 7070,D 9020,D 9380,U 9520,U 9670,U 9830,D 10280,D 11270,D 11390,U 13530,U 13610,D 14930,D 16120,D 16920,D 17620,D
NSTRSS	INTEGER			1040,D 1330,S 1790,U 5740,D 6650,D 6910,U 6920,U 6920,U 6930,U 13630,D
NT	INTEGER			120,D 1110,D 1230,S 1440,U 1450,U 1490,U 1530,U 1590,U 1610,U 1640,U 1660,U 1690,U 1730,U 1740,U 1790,U 2650,U 2680,U 2720,U 2740,U 2770,U 3080,U 3090,U 3130,U 3140,U 3270,U 3280,U 3300,U 3320,U 4900,U 4910,U 4930,U 5000,U 10970,S 10980,U 14970,D 16170,D
NTF	INTEGER			3260,S 3290,U 4890,S 4920,U
NTHETA	INTEGER			1040,D 1340,S 1340,U 1810,U 1810,U 1930,U 5740,D 6050,U 6650,D 13630,D 13820,U
NW	INTEGER			11270,D
P	REAL*8	74		14890,D 15080,S 15090,U 15170,S 15230,S 15230,U 15270,U 15310,U 15480,U 15520,U 15570,U 15760,U 16140,D 16300,S 16360,U 16360,S 16390,U 16530,U 16610,U 16670,U
PAV	REAL*8			14580,S 14610,U 14630,U
PH	REAL*8	50		1030,D 2720,S 2740,S 2790,U 2800,U 3030,U 6610,D 7140,D 13650,D 14580,U 14580,U 16880,D
PHP	REAL*8	50		1030,D 2720,S 2740,S 3030,U 6610,D 7140,D 13650,D 14220,U 16880,D 17150,U 17150,U 17160,U 17160,U 17190,U 17190,U 17210,U 17210,U
PHPP	REAL*8			17170,S 17190,S 17210,S 17360,U 17390,U
PHPP1	REAL*8			17150,S 17170,U
PHPP2	REAL*8			17160,S 17170,U
PI	REAL*8			1910,S 1920,U 15020,S 15570,U 15580,U 15760,U 15770,U 15780,U 16230,S 16550,U 16560,U 16630,U 16640,U 17010,S 17020,U 17040,U 17060,U 17080,U
PINT	REAL*8			15480,S 15520,S 15570,S 15610,S 15680,U 15690,U 15760,S 15830,U 15840,U 16480,S 16530,S 16530,U 16550,U 16550,S 16610,U 16610,S 16630,S 16630,U 16670,S 16700,U 16730,U
PI02	REAL*8			17690,S 17950,U 17960,U 17970,U
PI04	REAL*8			17990,S 18440,U 18450,U 18460,U 18470,U

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED											
PRINT	REAL*8			80,D	1050,D	5750,D	6660,D	11290,D							
PS	REAL*8			1000,D	9040,D	9850,D									
Q	REAL*8		8	14960,D 15680,S 15810,S 17320,S	15410,S 15680,U 15820,S 17340,S	15640,U 15690,S 15830,S 17360,S	15640,S 15690,U 15840,S 17380,S	15650,S 15700,U 15850,S 17400,S	15650,U 15700,S 15860,S 17420,S	15660,S 15710,S 15860,S 17470,S	15660,U 15710,U 15860,S 17470,S	15670,S 15710,U 15860,S 17510,U	15670,S 15800,S 17280,S 17510,U	15670,U 15800,S 17300,S	
QB3	REAL*8			14120,S	14200,U	14210,U									
QB7	REAL*8			14130,S	14200,U	14210,U									
QDC1	REAL*8			4970,S	4990,U										
QDC2	REAL*8			4960,S	4990,U										
QDC3	REAL*8			4950,S	4960,U	4970,U	4990,U								
QLOAD	REAL*8		204	940,D 9330,U 10030,U	5680,D 9330,S 10140,S	6070,S 9350,S 10210,U	6180,U 9350,U	6180,S 9490,U	6200,S 9500,S	6200,U 9650,U	6270,U 9790,D	8980,D 9960,P	9280,P 10030,S		
QLOAD1	REAL*8		1020	9070,D	9170,P	9220,U	9220,S	9240,S	9240,U	9330,S	9350,S	9500,S			
QN	REAL*8		1020	30,D 2590,U 6200,U 7820,U 7850,U 9200,S 14120,U 16940,D	140,S 2590,U 6330,U 7820,U 7850,U 9280,P 14130,U	950,D 3320,S 6580,D 7830,U 7860,U 9490,S	2030,S 3380,U 7200,D 7830,U 7860,U 9650,U	2330,S 3390,U 7800,U 7830,U 7860,U 9800,D	2340,S 4990,U 7800,U 7840,U 7870,U 9940,U	2350,S 5000,U 7800,U 7840,U 7870,U 10050,U	2360,S 5690,D 7800,U 7840,U 7870,U 10210,S	2590,U 5980,U 7820,U 7850,U 8990,D 13690,D	2590,U 6180,U 7820,U 7850,U 9200,U 14120,U 14910,D		
QN1	REAL*8		1020	30,D 2600,U 9200,U	950,D 3320,S 9650,U	2040,S 3390,U 9800,D	2470,S 3400,S 9940,U	2480,S 4990,U 10040,U	2490,S 5690,D 10050,S	2500,S 6580,D 13690,D	2600,U 7200,D 14910,D	2600,U 8990,D 16940,D	2600,U 9170,P		
QN2	REAL*8		1020	30,D 9940,S	950,D 9960,P	4990,U 10040,S	5690,D 13690,D	6580,D 14910,D	7200,D 16940,D	8990,D	9650,S	9800,D	9940,U		
QP	REAL*8		1020	30,D 6740,S 13690,D	950,D 6990,S 14910,D	3420,U 6990,U 16940,D	3430,S 7000,S	3430,U 7200,D	3430,U 8990,D	4990,S 9350,U	5000,U 9800,D	5690,D 10010,U	6580,D 10060,U		
QPR	REAL*8		8,5	6630,D 8250,U 8590,S 8860,S 8910,S	6990,U 8260,S 8600,S 8870,U	7000,U 8260,U 8600,U 8870,S	7120,D 8280,U 8610,S 8880,S	7990,S 8280,S 8610,U 8880,U	8210,S 8290,S 8620,U 8890,S	8210,U 8300,U 8620,S 8890,U	8230,S 8580,U 8630,S 8900,U	8230,U 8580,S 8630,U 8900,S	8240,S 8590,U 8360,U 8910,U		
QPRIME	REAL*8			6870,U	7050,D										

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VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
QP1	REAL*8		1020	30,D 520,S 950,D 3320,S 3410,U 3420,S 5000,U 5690,D 6580,D 7200,D 8990,D 9240,U 9800,D 10010,U 10060,S 13690,D 14910,D 16940,D
QQ	REAL*8		8	14960,D 15890,S 15910,S 15910,U 15940,U 16960,D 17490,S 17510,S 17510,U 17540,U
QS	REAL*8			30,D 950,D 5690,D 6580,D 7200,D 8990,D 9800,D 13690,D 14910,D 16940,D
QSS	REAL*8			10010,S 10030,U
QUES	REAL*8			14960,D 16960,D
Q1	REAL*8			2300,S 2330,U 2440,S 2470,U
Q2	REAL*8			2300,S 2340,U 2440,S 2480,U
Q3	REAL*8			2300,S 2350,U 2440,S 2490,U
Q4	REAL*8			2300,S 2360,U 2440,S 2500,U
R	REAL*8		50	1030,D 2720,S 2740,S 3020,U 6610,D 7140,D 11310,D 11430,S 11580,S 11580,U 11580,U 11590,U 11590,U 11590,S 11590,U 11600,U 11600,S 11600,U 11600,U 11600,U 11600,U 12270,U 12270,S 12270,U 12270,U 12270,U 12270,U 12360,U 12360,U 12360,U 12360,U 12360,U 12360,S 12360,U 12460,U 12460,U 12460,U 12460,U 12460,U 12460,S 12460,U 12460,U 12470,U 12570,S 12570,U 12570,U 12570,U 12570,U 12570,U 12570,U 12570,U 12570,U 12580,U 12580,U 13060,U 13060,S 13060,U 13070,U 13070,S 13070,U 13070,U 13080,S 13080,U 13080,U 13080,U 13080,U 13120,U 13120,U 13120,U 13120,U 13120,U 13120,U 13120,U 13120,S 13130,U 13130,U 13130,U 13130,S 13130,U 13130,U 13140,U 13150,U 13150,S 13150,U 13150,U 13150,U 13150,U 13150,U 13160,U 13160,U 13170,U 13170,U 13170,U 13170,U 13170,U 13170,U 13170,U 13170,U 13180,U 13180,U 13240,U 13240,S 13250,U 13250,U 13250,S 13260,U 13260,U 13260,U 13260,S 13270,U 13270,U 13270,U 13270,S 13270,U 13270,U 13320,U 13320,U 13320,U 13320,U 13320,U 13340,S 13340,S 13340,U 13340,U 13340,U 13340,U 13340,U 13350,U 13360,S 13360,U 13360,U 13360,U 13360,U 13360,U 13370,U 13370,U 13380,U 13380,U 13380,U 13380,S 13380,U 13380,U 13390,U 13390,U 13390,U 13390,U 13410,S 13410,U 13410,U 13410,U 13410,U 13410,U 13430,S 13430,U 13430,U 13430,U 13430,U 13430,U 13440,U 13440,U 13450,U 13450,U 13450,S 13450,U 13450,U 13450,U 13460,U 13460,U 13470,S 13470,U 13470,U 13470,U 13470,U 13470,U 13480,U 13480,U 13480,U 13510,U 13650,D 14170,U 14210,U 14220,U 14220,U 14570,U 14570,U 14600,U 14600,U 14620,U 14620,U 14890,D 15080,S 15090,U 15170,S 15240,S 15240,U 15270,U 15310,U 15490,U 15530,U 15580,U 15770,U 16140,D 16300,S 16370,S 16370,U 16390,U 16540,U 16620,U 16680,U 16880,D
RAD	REAL*8			1920,S 1940,U
RAV	REAL*8			14570,S 14600,U 14620,U
RESFRT	REAL*8			100,D 1070,D 5770,D 9060,D
RINT	REAL*8			15490,S 15530,S 15580,S 15620,S 15640,U 15650,U 15660,U 15670,U 15770,S 15790,S 1580,U 15810,U 15820,U 16490,S 16540,U 16540,S 16560,U 16560,S 16620,S 16620,U 16640,S 16640,U 16680,S 16710,S 16740,U
RL	REAL*8			8670,S 8860,U 8870,U 8880,U 8890,U 8900,U 8910,U

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
RM	REAL*8			7360,S 7370,U 7900,U 8670,U
RSL	REAL*8			7900,S 8210,U 8230,U 8250,U 8260,U 8280,U 8300,U 8580,U 8590,U 8600,U
RSTRNT	REAL*8			970,D 9710,D 9820,D 10300,D
RZ	REAL*8			1120,D 7220,D
RQ	REAL*8		51	1120,D 2770,S 7220,D 7310,U 7310,U 7360,U 7360,U
R2I	REAL*8			7370,S 7400,U 7410,U 7420,U 7480,U 7490,U 7550,U 7560,U 7570,U 7580,U
S	REAL*8		74	14890,D 16140,D 15080,S 15090,U 15170,S 15250,U 15250,S 15270,U 15310,U 15540,U 15780,U
SCCS	REAL*8		625	7100,D 8540,U 17610,D 18470,U
SETUP	REAL*8			690,U 5660,D
SHRS	REAL*8			14600,S 14640,U
SHRT	REAL*8			14620,S 14650,U
SINE	REAL*8		51	1020,D 2720,S 2740,S 2740,S 6600,D 7130,D 13640,D 14120,U 14130,U 16870,D
SINM	REAL*8		50	1030,D 2800,S 6610,D 7140,D 13650,D 14170,U 14200,U 14200,U 14220,U 16880,D
SINT	REAL*8			15500,S 15540,S 15590,S 15630,S 15700,U 15710,U 15780,S 15850,U 15860,U
SIPH	REAL*8			7340,S 7440,U 7460,U 7480,U 7490,U 7560,U 7580,U
SLVEEQ	REAL*8			940,D 5680,D 8980,U 9790,D 10270,D 11260,D
SL2R	REAL*8			7560,S 7640,U
SN	REAL*8			6110,S 6200,U 13990,S 14060,U 14080,U 14240,U 14250,U 14260,U
SOLVEQ	REAL*8			9450,U 10180,U 11240,D
SC2R	REAL*8			7580,S 7620,U
SPA	REAL*8		6550	11260,D 11410,U 13550,S
SPR	REAL*8		204	11260,D 11430,U 13510,S
SSC	REAL*8		125	7090,D 8100,U 8120,U 8150,U 17600,D 17960,S
SSCC	REAL*8		625	7100,D 8520,U 17610,D 18460,S
SSSS	REAL*8		625	7100,D 8540,U 17610,D 18450,S



VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HDW	USED									
STIFM	REAL*8		6550	10270,D 10850,U 11190,U	10450,S 10870,D 11210,U	10490,S 11040,U	10510,S 11050,U	10580,S 11060,U	10630,S 11060,U	10650,S 11070,U	10710,S 11070,U	10790,S 11070,U	10820,S 11180,U	
STRESS	REAL*8			6930,U	13580,D									
STRMS	REAL*8			14370,S	14430,U	14460,U	14540,U	14600,U	14640,U	14660,U				
STRMST	REAL*8			14390,S	14450,U	14480,U	14540,U	14620,U	14630,U	14640,U	14680,U			
STRMT	REAL*8			14380,S	14440,U	14470,U	14540,U	14610,U	14640,U	14670,U				
STRNS	REAL*8			14340,S	14430,U	14460,U	14540,U	14640,U						
STRNST	REAL*8			14360,S	14450,U	14480,U	14540,U	14640,U						
STRNT	REAL*8			14350,S	14440,U	14470,U	14540,U	14640,U						
STTMST	REAL*8			14510,S	14600,U	14690,U								
STTRMT	REAL*8			14500,S	14620,U	14700,U								
T	REAL*8		50	1020,D 7130,D	2690,S 13640,D	3030,U 14410,U	6600,D 14420,U	6780,U 16870,D	6790,U 17020,U	6800,U 17040,U	6810,U 17060,U	6820,U 17080,U	6830,U	
TAPES	REAL*8			120,D	1110,D	14970,D	16170,D							
TDT2	REAL*8			9290,S	9330,U	9350,U								
TEST	REAL*8	END		150,S	310,U	330,U	400,U							
TFORCE	REAL*8			4520,U	16850,D									
TH	REAL*8		50,5,2	1080,D 4580,U 14010,U 17360,U	3300,S 4770,S 14010,U 17380,U	3630,S 4770,U 16150,D 17400,U	3640,S 4930,U 16730,S 17420,U	3770,U 7170,D 16890,D 17270,U	3770,S 8790,U 17280,U 17300,U	3890,U 8790,U 17300,U 17320,U	3890,S 8790,U 17300,U 17320,U	3910,S 13710,D 17320,U 17340,U	4460,S 14010,U 17340,U	
THCOE	REAL*8			4510,U	16100,D									
THCON	REAL*8			1090,D	7180,D									
THER	REAL*8			1680,D	7170,D	13710,D	16150,D	16890,D						
THETA	REAL*8			1040,D 13630,D	1340,D 13980,U	1810,S 13990,U	1940,S 14520,U	1940,U	5740,D	6110,U	6120,U	6230,U	6650,D	
THETAS	REAL*8			1040,U	5740,D	6650,D	13630,D							
THETA1	REAL*8			6230,S	6240,U	14520,S	14540,U	14640,U						

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
THETB	REAL*8		74	14890,D 15170,S 15210,U 15220,S 15270,U 15280,U 15310,U 15320,U 15340,U 15340,S 15450,U 15460,U 15480,U 15480,U 15490,U 15490,U 16140,D 16300,S 16330,U 16350,S 16390,U 16390,U 16430,U 16430,S 16530,U 16530,U 16540,U 16540,U 16590,U 16600,U
THT	REAL*8			8790,S 8820,U 8830,U 14010,S 14090,U 14100,U
TH1	REAL*8		5	1130,D 4420,S 4460,U
TIM	REAL*8			3990,S 4050,U
TIME	REAL*8			70,D 630,S 640,U 690,P 730,U 730,S 1010,D 4910,U 5010,U 5660,U 5930,U 6350,U 6570,D 7190,D 8790,U 9050,D 9140,U 9150,U 9860,D 9980,U 13680,D 13790,U 14010,U 14020,U
TIMEP	REAL*8			100,D 560,S 580,U 630,U 1070,D 3280,S 3500,U 5770,D 9060,D
TMFT	REAL*8			70,D 1010,D 6570,D 7190,D 9050,D 9860,D 13680,D
TM1	REAL*8			13790,S 13800,U
TOTIME	REAL*8			70,D 580,U 1010,D 1320,S 1780,U 3800,U 4000,U 6570,D 7190,D 9050,D 9860,D 13680,D
TPRINT	REAL*8			13800,S 13810,U
TPRNT	REAL*8			3500,S 3510,U 4050,S 4060,U 5010,S 5020,U 5930,S 5940,U 6220,U
TRI4OR	REAL*8			3210,U 17580,D
TO	REAL*8			70,D 570,S 660,S 1010,D 3280,S 3790,S 3960,S 4910,U 6570,D 7190,D 8790,U 8790,U 9050,D 9140,U 9140,U 9150,U 9150,U 9860,D 9980,U 9980,U 13680,D 14010,U 14010,U 14020,U 14020,U
T1	REAL*8			70,D 640,U 660,U 1010,D 3280,S 3790,U 3800,S 3960,U 3980,S 3990,U 4000,S 4910,U 6570,D 7190,D 8790,U 9050,D 9140,U 9150,U 9860,D 9980,U 13680,D 14010,U 14020,U
T3	REAL*8			9140,S 9350,U 9990,S 10010,U
T3M1	REAL*8			9150,S 9240,U
XIHL	REAL*8			6100,S 6110,U 6120,U 13970,S 13980,U 13990,U 14170,U 14200,U 14200,U 14210,U 14250,U 14260,U 14270,U
XK	REAL*8			7610,S 7620,U 7630,U 7640,U 7650,U 7700,U
XKEEP	REAL*8			3380,S 3400,U 3410,S 3430,U 9540,S 9560,U 9690,S 9710,U
XN	REAL*8	6550		940,U 1150,S 1150,S 3130,S 3170,U 5680,D 8980,D 9170,P 9400,U 9400,S 9540,U 9550,S 9690,U 9700,U 9700,S 9790,D
XP	REAL*8	6550		1000,D 3140,S 3190,U 9040,D 9280,P 9400,U 9550,U 9560,S 9700,U 9710,S 9850,D 9960,P

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
X1	REAL*8			15450,S 15520,U 15530,U 15540,U 16590,S 16610,U 16620,U
X2	REAL*8			15460,S 15520,U 15530,U 15540,U 16600,S 16610,S 16620,S
YKP	REAL*8			15380,S 15450,U 15460,U 15520,U 15530,U 15540,U 16470,S 16590,U 16600,U 16610,U 16620,U 17260,S 17280,U 17300,U 17330,U 17350,U 17400,U 17400,U 17420,U 17420,U
Z	REAL*8		51	1120,D 2770,S 7220,D 7320,U 7320,U

FRCFS

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
AI	REAL*8		167	14070,D 15640,U 15650,U 15660,U 15670,U 15680,U 15690,U 15700,U 15710,U 15790,U 15870,U 15810,U 15820,U 15830,U 15840,U 15850,U 15860,U
ALPHK	REAL*8			14870,U
ANG	REAL*8			15200,S 15220,U
CHALS	REAL*8			14970,D
CHECK	REAL*8		8,8	14970,D 15910,U
CONST	REAL*8			14930,D
C12	REAL*8			15740,S 15750,S 15760,U 15770,U 15780,U
DCOS	REAL*8			15540,U 15540,U
DSIN	REAL*8			15520,U 15520,U 15530,U 15530,U
DT2	REAL*8			14930,D
FORCE	REAL*8		2040	14910,D 15940,S 15940,U
FRCF	REAL*8			14990,D
FRCFS	REAL*8			14870,D
HARM	REAL*8			14950,D
I	INTEGER			15080,U 15080,U 15080,U 15080,S 15090,U 15090,U 15090,U 15090,U 15090,U 15090,S 15170,U 15170,U 15170,U 15170,U 15170,S 15270,U 15270,U 15270,U 15270,U 15280,U 15280,S 15280,U 15310,U 15310,U 15310,U 15320,U 15320,U 15320,S 15400,S 15410,U 15430,U 15450,U 15460,U 15480,U 15480,U 15480,U 15490,U 15490,U 15490,U 15520,U 15530,U 15540,U 15880,S 15890,U 15910,U 15910,U 15910,U 15920,S 15930,U 15940,U
IP	INTEGER			14870,U 15930,U
IDCOF	INTEGER			14940,D 15050,U 15390,U
IDELF	INTEGER			14940,D
IDP	INTEGER			15330,S 15340,U 15340,U
IFLM	INTEGER			14870,U 15030,U 15040,U 15040,U 15060,U 15130,U 15930,U
IFLM1	INTEGER			15080,S 15090,U 15170,S 15270,U
IFLM2	INTEGER			15070,S 15040,U 15080,S 15090,U 15170,S 15270,U
IH	INTEGER			15360,S 15370,U 15760,U 15770,U 15780,U 15930,U
IHARM	INTEGER			14950,D 15370,U

FRCS

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
J	INTEGER			15000,S 15010,U 15010,U 15030,S 15940,U 15940,U
KEY	INTEGER			15210,S 15220,U 15230,U 15240,U 15250,U
KYP	INTEGER			15370,S 15380,U 15470,U 15560,U 15750,U
LF	INTEGER			15040,U
N	INTEGER			14930,D
ND	INTEGER			14070,D 15080,U 15170,U
NDP	INTEGER			15170,S 15170,U 15180,U 15190,U 15280,U 15290,U 15300,U 15310,U 15420,U 15440,U
NDP02	INTEGER			15300,S 15320,U
NDP1	INTEGER			15420,S 15430,U
NDP2	INTEGER			15180,S 15210,U
ND2	INTEGER			15200,S 15320,U 15330,U
NELFMS	INTEGER			14930,D
NFO	INTEGER			14930,D 15930,U
NEQT	INTEGER			14930,D 15930,U
NH	INTEGER			14930,D 15080,U 15090,U 15360,U
NHNS	INTEGER			14930,D
NHP	INTEGER			14950,D
NN	INTEGER			14930,D
NN00FS	INTEGER			14930,D
NPRNTE	INTEGER			14940,D
NPRNT1	INTEGER			14940,D 15060,U 15090,U 15130,U 15270,U 15310,U
NPRNTQ	INTEGER			14930,D
NS	INTEGER			14970,D
NSIZE	INTEGER			14930,D
NT	INTEGER			14970,D

FRCFS

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
P	REAL*8		74	14890,D 15080,S 15090,U 15170,S 15230,S 15230,U 15270,U 15310,U 15480,U 15520,U 15570,U 15760,U
PT	REAL*8			15020,S 15570,U 15580,U 15760,U 15770,U 15780,U
PINT	REAL*8			15480,S 15520,S 15570,S 15610,S 15680,U 15690,U 15760,S 15830,U 15840,U
Q	REAL*8		8	14960,D 15410,S 15640,S 15640,U 15650,S 15650,U 15660,S 15660,U 15670,S 15670,U 15680,S 15680,U 15690,S 15690,U 15700,S 15700,U 15710,S 15710,U 15790,S 15800,S 15810,S 15820,S 15830,S 15840,S 15850,S 15860,S 15910,U
QV	REAL*8		1020	14910,D
QV1	REAL*8		1020	14910,D
QV2	REAL*8		1020	14910,D
QP	REAL*8		1020	14910,D
QP1	REAL*8		1020	14910,D
QQ	REAL*8		8	14960,D 15890,S 15910,S 15910,U 15940,U
QS	REAL*8			14910,D
QIFC	REAL*8			14960,D
R	REAL*8		74	14890,D 15080,S 15090,U 15170,S 15240,S 15240,U 15270,U 15310,U 15490,U 15530,U 15580,U 15770,U
RINT	REAL*8			15400,S 15530,S 15580,S 15620,S 15640,U 15650,U 15660,U 15670,U 15770,S 15790,S 15800,U 15810,U 15820,U
S	REAL*8		74	14890,D 15080,S 15090,U 15170,S 15250,S 15250,U 15270,U 15310,U 15540,U 15780,U
SINT	REAL*8			15520,S 15540,S 15590,S 15630,S 15700,U 15710,U 15780,S 15850,U 15860,U
TAPES	REAL*8			14970,D
THETB	REAL*8		74	14890,D 15170,S 15200,U 15220,S 15270,U 15280,U 15310,U 15320,U 15340,S 15340,U 15450,U 15460,U 15480,U 15480,U 15490,U 15490,U
X1	REAL*8			15450,S 15520,U 15530,U 15540,U
X2	REAL*8			15460,S 15520,U 15530,U 15540,U
YKP	REAL*8			15380,S 15450,U 15460,U 15520,U 15530,U 15540,U

HOURON

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
CONST	REAL*8			9830,D
DELTE	REAL*8			9860,D
DT2	REAL*9			9830,D 10030,U
FORCE	REAL*8		2040	9820,D 10010,U 10010,U 10010,U
HOURON	REAL*8			9770,D
I	INTEGER			9920,S 9930,U 9920,S 10000,U 10030,U 10030,U 10120,S 10130,U 10190,S 10200,U 10210,U
INDOE	INTEGER			9840,D
INDEL	INTEGER			9840,D
IF0	INTEGER			10130,S 10140,U
IH	INTEGER			9770,U 9960,P 10180,P
KY	INTEGER			9770,U 10100,U
LK	INTEGER		204	9820,D 10130,U
MATMUT	INTEGER			9960,U
N	INTEGER			9930,D 9930,U 10000,U 10200,U
NCLOSE	INTEGER			9820,D 10100,U
NFLMS	INTEGER			9830,D
NE0	INTEGER			9830,D 9920,U 9960,P 9990,U 10190,U
NEQT	INTEGER			9830,D 10010,U
NH	INTEGER			9830,D
NHNS	INTEGER			9830,D
NN	INTEGER			9820,D
NNODES	INTEGER			9830,D
NODRE	INTEGER			10000,S 10100,S 10100,U 10110,U 10120,U
NODRES	INTEGER			9820,D 10090,U
NPI	INTEGER			9930,S 9940,U 9940,U 9940,U 9940,U 9940,U 10000,S 10010,U 10010,U 10010,U 10010,U 10010,U 10040,U 10040,U 10050,U 10050,U 10060,U 10060,U 10200,S 10210,U

HOURON

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
NPRNTE	INTEGER			9840,D
NDRNTI	INTEGER			9840,D
NDRNTD	INTEGER			9830,D
NSIZE	INTEGER			9830,D
PS	REAL*8			9850,D
QLOAD	REAL*8		204	9790,D 9960,P 10030,S 10030,U 10140,S 10210,U
QV	REAL*8		1020	9800,D 9940,U 10050,U 10210,S
QV1	REAL*8		1020	9800,D 9940,U 10040,U 10050,S
QV2	REAL*8		1020	9800,D 9940,S 9940,U 9960,P 10040,S
QP	REAL*8		1020	9800,D 10010,U 10060,U
QP1	REAL*8		1020	9800,D 10010,U 10060,S
QS	REAL*8			9800,D
QSS	REAL*8			10010,S 10030,U
RSTRVT	REAL*8			9820,D
SLVFFD	REAL*8			9790,D
SOLVED	REAL*8			10180,U
TIME	REAL*8			9860,D 9980,U
TMET	REAL*8			9860,D
TOTIME	REAL*8			9860,D
T0	REAL*8			9860,D 9980,U 9980,U
T1	REAL*8			9860,D 9980,U
T3	REAL*8			9980,S 10010,U
XN	REAL*8		6550	9790,D
XP	REAL*8		6550	9850,D 9960,P



HOUR01

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
CONST	REAL*8			9020,D
DELTF	REAL*8			9050,D 9150,U 9200,U
DELTFD	REAL*8			9060,D
DT2	REAL*8			9020,D 9290,U 9320,U 9350,U 9350,U 9400,U 9650,U
ENDCF	REAL*8		2047	8990,D 9220,U 9240,U 9240,U 9240,U 9330,U 9350,U 9360,U 9360,U
HOUR01	REAL*8			8960,D
I	INTEGER			9180,S 9190,U 9220,U 9220,U 9240,U 9240,U 9300,S 9310,U 9330,U 9330,U 9330,U 9330,U 9350,U 9350,U 9350,U 9380,S 9390,U 9470,S 9480,U 9490,U 9500,U 9520,S 9530,U 9630,S 9640,U 9650,U 9670,S 9680,U
IDCDE	INTEGER			9030,D
IDELF	INTEGER			9030,D
IFLAG	INTEGER			9130,S 9460,U 9590,S
IH	INTEGER			8960,U 9170,P 9280,P 9450,P
IPSTRT	INTEGER			9060,D 9210,U 9320,U
ITP	INTEGER			9060,D
KV	INTEGER			8960,U 9430,P 9740,P
LK	INTEGER		204	9010,D
MATMIJ	INTEGER			9170,U 9280,U
N	INTEGER			9020,D 9190,U 9310,U 9480,U 9640,U
NCLOSE	INTEGER			9010,D
NLEMS	INTEGER			9020,D
NFQ	INTEGER			9020,D 9170,P 9180,U 9280,P 9300,U 9470,U 9630,U
NFOT	INTEGER			9020,D 9240,U 9360,U
NH	INTEGER			9020,D
NHNS	INTEGER			9020,D
NN	INTEGER			9020,D 9390,U 9530,U 9680,U
NNODES	INTEGER			9020,D

HOURS

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
NNPI	INTEGER			9390,S 9400,U 9400,U 9400,U 9530,S 9540,U 9550,U 9550,U 9560,U 9680,S
NNDPES	INTEGER			9690,U 9700,U 9700,U 9700,U 9710,U
NPI	INTEGER			9010,D
NPRNIT	INTEGER			9190,S 9200,U 9200,U 9200,U 9220,U 9240,U 9240,U 9240,U 9250,U 9310,S
NPRNT	INTEGER			9320,U 9350,U 9350,U 9360,U 9360,U 9480,S 9490,U 9640,S 9650,U 9650,U
NPRNTE	INTEGER			9060,D
NPRNTL	INTEGER			9060,D
NPRNTD	INTEGER			9030,D
NPRNTL	INTEGER			9030,D
NPRNTD	INTEGER			9020,D
NPESTR	INTEGER			9420,U 9740,U
NVSIZE	INTEGER			9020,D 9380,U 9520,U 9670,U
PS	REAL*8			9040,D
Q1Q10	REAL*8		204	8990,D 9290,P 9330,S 9330,U 9350,S 9350,U 9490,U 9500,S 9650,U
Q1QAD1	REAL*8		1020	9070,D 9170,P 9220,S 9220,U 9240,S 9240,U 9330,S 9350,S 9500,S
QV	REAL*8		1020	8990,D 9200,S 9200,U 9280,P 9490,S 9650,U
QV1	REAL*8		1020	8990,D 9170,P 9200,U 9650,U
QV2	REAL*8		1020	8990,D 9650,S
QP	REAL*8		1020	8990,D 9350,U
QP1	REAL*8		1020	8990,D 9240,U
QS	REAL*8			8990,D
PESTRY	REAL*8			9060,D
RSTRNT	REAL*8			9010,D
SI VEEQ	REAL*8			8980,D
SOLVFO	REAL*8			9450,U
TOT2	REAL*8			9290,S 9330,U 9350,U
TIME	REAL*8			9050,D 9140,U 9150,U

HOUJAO1

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED									
TIMEP	REAL*8			9060,D									
TMET	REAL*8			9050,D									
TOTIME	REAL*8			9050,D									
T2	REAL*8			9050,D	9140,U	9140,U	9150,U	9150,U					
T1	REAL*8			9050,D	9140,U	9150,U							
T3	REAL*8			9140,S	9350,U								
T3M1	REAL*8			9150,S	9240,U								
XKEEP	REAL*8			9540,S	9560,U	9600,S	9710,U						
XV	REAL*8		6550	8980,D	9170,P	9400,S	9400,U	9540,U	9550,S	9690,U	9700,S	9700,U	
XP	REAL*8		6550	9040,D	9280,P	9400,U	9550,U	9560,S	9700,U	9710,S			

INPUT

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
AI	REAL*8		167	930,D 2650,S 2660,U 4310,S 4490,S
ALPHK	REAL*8			4320,D
ALS	REAL*8		50	1080,D 2890,S 2940,S 3020,U
ALSTI	REAL*8			2870,S 2890,U
ALT	REAL*8		50	1080,D 2900,S 2950,S 3020,U
ALTI	REAL*8			2870,S 2900,U
ARCL	REAL*8		50	1030,D 2720,S 2740,S 3030,U
CHALS	REAL*8			920,D
CHECK	REAL*8		8,8	930,D 2650,S 2660,U 4310,S 4490,S
COMMENT	REAL*8		20	1130,D 1150,S 1270,S 1280,U 1400,S 1510,U
CONST	REAL*8			980,D
CONSTE	REAL*8			1160,D 1200,S 3710,U 3980,S 4000,U 4060,U 4720,U
CONSTN	REAL*8	CONSTANT		1160,D 1170,S 3710,U 4000,U 4720,U
CONST1	REAL*8	R * S		1160,D 1170,S 1260,U
COSINE	REAL*8		51	1030,D 2720,S 2740,S 2740,S
COSM	REAL*8		50	1030,D 2790,S
CYCLE	REAL*8			1100,D
DATA	REAL*8			1170,S
DCOS	REAL*8			2790,U
DELTE	REAL*8			1010,D
DELTEP	REAL*8			1070,D 3280,S 3430,U 3510,U
DELTEF	REAL*8			1320,S 1780,U 2010,U 3430,U 4910,U 4950,U
DOUBLE	REAL*8			1160,D
DSIN	REAL*8			2800,U
DTH	REAL*8		50,5,2	1080,D 3310,S 3650,S 3660,S 3780,S 3780,U 3900,S 3900,U 3920,S 4450,S 4580,U 4780,S 4780,U 4940,U
DTH1	REAL*8		5	1130,D 4420,S 4450,U



INPUT

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
IB	INTEGER			3040,S 3050,S 3060,U 4000,U 4100,U 4320,P 4390,U 4510,P 4520,P 4650,U 4720,U 4820,S 4820,U 4830,U
IRP1	INTEGER			3300,U 3300,S 3310,U 3310,S 4390,S 4450,U 4460,U 4580,U 4580,U 4930,U 4930,S 4940,U 4940,S
IDCOF	INTEGER			000,D 3980,S
IDELF	INTEGER			000,D 3990,S 4280,U
IFLM	INTEGER			2880,S 2890,U 2900,U 2930,S 2940,U 2950,U 3300,U 3300,S 3310,U 3310,S 4200,S 4320,D 4430,S 4450,U 4460,U 4480,S 4510,P 4520,P 4570,S 4580,U 4580,U 4580,U 4930,U 4930,S 4940,U 4940,S
IFLM1	INTEGER			2870,S 2880,U 4420,S 4430,U
IFLM2	INTEGER			2870,S 2880,U 2910,U 4420,S 4430,U 4470,U
IFLAG	INTEGER			2320,S 2330,U 2340,U 2350,U 2360,U 2460,S 2470,U 2480,U 2490,U 2500,U
IH	INTEGER			2270,S 2280,U 2410,S 2420,U 3040,S 3060,U 3300,U 3300,S 3310,U 3310,S 4000,S 4000,U 4140,U 4420,U 4420,U 4420,S 4440,S 4450,U 4460,U 4460,U 4460,U 4550,S 4560,U 4580,U 4580,U 4640,S 4650,U 4660,U 4930,U 4930,S 4940,U 4940,S
IHARM	INTEGER		5	1060,D 1370,U 1740,U 1800,U 2590,U 3060,U 3160,U 3180,U 3270,S 4140,U 4560,U 4660,U 4900,U
IHI	INTEGER			4090,S 4180,U
II	INTEGER			2570,S 2580,U 2590,U 2640,S
IN	INTEGER			4170,S 4180,U 4230,U
INCRST	INTEGER			1320,S 1780,U 3350,U
INODE	INTEGER			2310,S 2320,U 2450,S 2460,U
INPUT	INTEGER			010,D
IN1	INTEGER			2300,S 2310,U 2440,S 2450,U 4160,S 4170,U
IN2	INTEGER			2300,S 2310,U 2370,U 2440,S 2450,U 2510,U 4160,S 4170,U 4250,U
IPRINT	INTEGER			1050,D 1320,S 1780,U
IQ	INTEGER			2560,S 2590,U 2590,U 2590,U 2590,U 2600,U 2600,U 2600,U 2600,U
IQN	INTEGER			2250,S 2260,U
IQN1	INTEGER			2250,S 2400,U

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INPUT

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
TRSTRT	INTEGER			1070,D 1320,S 1370,U 1390,U 1780,U 2060,U 3220,U
ITAM	INTEGER			1100,D 4910,U 5020,U
ITCOF	INTEGER			1090,D 3980,S 4400,U 4500,U
ITFLF	INTEGER			1090,D 1320,S 1790,U 2820,U 4340,U
ITP	INTEGER			1070,D 3280,S 3350,U 3510,U
IX	INTEGER			2580,S 2590,U 2590,U 2590,U 2600,U 2600,U 2600,U 2600,U 2600,U
J	INTEGER			1270,U 1270,S 1280,U 1280,S 1490,U 1490,S 1510,U 1510,S 1600,U 1610,U 1680,S 1690,U 1700,S 1720,U 2650,U 2650,S 2660,U 2660,S 3620,S 3630,U 3640,U 3650,U 3660,U 3760,S 3770,U 3770,U 3780,U 3780,U 3880,S 3890,U 3900,U 3900,U 3900,U 3910,U 3920,U 4310,U 4310,S 4490,U 4490,S 4760,S
JH	INTEGER			3050,S 3060,U 3120,U 3160,U 3180,U
JUNK	INTEGER		20	1130,D 1450,S 1530,S 1740,S 3270,S 4900,U
K	INTEGER			1480,S 1580,S 1620,S 1630,U 1720,S 4180,S 4190,U 4200,U 4210,U 4220,U 4690,S 4700,U 4700,U 4700,U 4700,U
KFY	INTEGER			910,U 1180,U 3710,U 3950,U
KEYRS	INTEGER			1380,S 1390,S 1500,U 1540,U 1750,S
KK	INTEGER			4650,S 4690,U
KYP	INTEGER			4660,S 4670,U
LK	INTEGER		204	970,D 2180,S 2190,S 2200,S 3280,S 3330,S 3340,S 4910,U
LL	INTEGER			1050,D
N	INTEGER			980,D 2280,S 2320,U 2420,S 2460,U
NCARDS	INTEGER			1230,S 1240,U 1250,U 1450,S 1460,U 1480,U
NCF	INTEGER			3980,S 4070,U
NCF1	INTEGER			4110,S 4120,U
NCLCST	INTEGER			1040,D 1330,S 1790,U
NCLOSE	INTEGER			970,D 1320,S 1790,U
ND	INTEGER			1110,D 1230,U 1270,U 1320,U 1340,U 1370,U 1790,U 2110,U 2160,U 2250,U 2300,U 2440,U 2870,U 3980,U 4110,U 4160,U 4420,U

INPUT

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
NDIRCT	INTEGER			2160,S 2170,U 2180,U
NELFMS	INTEGER			980,D 1530,S 1580,U 1600,U 1620,U 1630,U 1680,U 1690,U 1710,U 1790,U 1960,U 1000,U 2640,U 2680,U 2680,U 2680,U 2690,U 2690,U 2700,U 2700,U 2710,U 2700,U 2710,U 2780,U 2910,U 2920,U 3020,U 3000,U 3310,U 3610,U 3750,U 3870,U 4300,U 4470,U 4480,U 4570,U 4750,U 4930,U 4940,U
NFO	INTEGER			980,D 1970,S 1980,U 2280,U 2420,U 2560,U 4180,U 4650,U
NFOT	INTEGER			980,D 1980,S 2020,U 3260,U 3320,U 3320,U 3320,U 3370,U 3570,U 3590,U 3730,U 3740,U 3830,U 3840,U 3850,U 4180,U 4650,U 4730,U 4740,U 4890,U 4980,U 5000,U 5000,U 5000,U
NH	INTEGER			980,D 1370,S 1370,U 1740,S 1740,U 1800,U 1800,U 1980,U 2000,U 2270,U 2410,U 2550,U 3050,U 3270,S 3270,U 3300,U 3310,U 3620,U 3760,U 3880,U 4080,U 4420,U 4440,U 4550,U 4640,U 4760,U 4900,U 4900,U 4930,U 4940,U
NHNS	INTEGER			980,D 2000,S
NHP	INTEGER			1060,D 1530,S 1700,U 3040,U
NLTERM	INTEGER			3360,U
NW	INTEGER			980,D 3120,S 3130,U 3140,U 3170,U 3190,U
NNODES	INTEGER			980,D 1960,S 1970,U 2370,U 2510,U 2570,U 2770,U 2770,U 4250,U 4680,U
NNDRES	INTEGER			970,D 2110,S 2120,U 2130,U 2140,U 2190,U 2200,U 3280,S 3280,U 3330,U 3340,U 4010,U 4910,U
NOUT	INTEGER			1050,D 1350,S
ND	INTEGER			2160,S 2170,U 2180,U
NDRNIT	INTEGER			1070,D 1330,S 1780,U
NDRNMS	INTEGER			1330,S 1800,U 3150,U
NDRNT	INTEGER			1070,D 1330,S 1780,U
NDRNTE	INTEGER			990,D 1330,S 1790,U 4010,U 4600,U
NDRNTH	INTEGER			1090,D 1330,S 1790,U 4540,U
NDRNTI	INTEGER			990,D 1330,S 1790,U 4010,U 4130,U 4230,U
NDRNTQ	INTEGER			980,D 1320,S 1780,U
NS	INTEGER			1110,D 1790,U 2630,U 2660,U 4290,U 4310,U 4380,U 4490,U



INPUT

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
NSIZE	INTEGER			980,D 1710,S 1720,U 1990,S 2000,U 3080,U 3090,U 3120,U 3130,U 3140,U 3170,U 3180,U
NSTRSS	INTEGER			1040,D 1330,S 1700,U
NT	INTEGER			1110,D 1230,S 1440,U 1450,U 1490,U 1530,U 1590,U 1610,U 1640,U 1660,U 1690,U 1730,U 1740,U 1790,U 2650,U 2680,U 2720,U 2740,U 2770,U 3080,U 3090,U 3130,U 3140,U 3270,U 3280,U 3300,U 3320,U 4900,U 4910,U 4930,U 5000,U
NTE	INTEGER			3260,S 3290,U 4890,S 4920,U
NTHETA	INTEGER			1040,D 1340,S 1360,U 1810,U 1810,U 1930,U
PH	REAL*8		50	1030,D 2720,S 2740,S 2790,U 2800,U 3030,U
PHD	REAL*8		50	1030,D 2720,S 2740,S 3030,U
PI	REAL*8			1910,S 1920,U
PPINT	REAL*8			1050,D
PS	REAL*8			1070,D
QDC1	REAL*8			4970,S 4990,U
QDC2	REAL*8			4960,S 4990,U
QDC3	REAL*8			4950,S 4960,U 4970,U 4990,U
QLOAD	REAL*8		204	940,D
QV	REAL*8		1020	950,D 2030,S 2320,S 2330,S 2340,S 2350,S 2360,S 2590,U 2590,U 2590,U 2590,U 3320,S 3380,U 3390,U 4990,U 5000,U
QV1	REAL*8		1020	950,D 2040,S 2470,S 2480,S 2490,S 2500,S 2600,U 2600,U 2600,U 2600,U 3320,S 3390,U 3400,S 4990,U
QV2	REAL*8		1020	950,D 4990,U
QP	REAL*8		1020	950,D 3420,U 3430,U 3430,S 3430,U 4990,S 5000,U
QP1	REAL*8		1020	950,D 3320,S 3410,U 3420,S 5000,U
QS	REAL*8			950,D
Q1	REAL*8			2370,S 2370,U 2440,S 2470,U
Q2	REAL*8			2370,S 2340,U 2440,S 2480,U
Q3	REAL*8			2300,S 2350,U 2440,S 2490,U

INPUT

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
Q4	REAL*8			2300,S 2360,U 2440,S 2500,U
Q	REAL*8		50	1030,D 2720,S 2740,S 3020,U
QAD	REAL*8			1020,S 1940,U
QFSTRY	REAL*8			1070,D
QSTRNY	REAL*8			970,D
Q7	REAL*8			1120,D
Q9	REAL*8		51	1120,D 2770,S
SINF	REAL*8		51	1020,D 2720,S 2740,S 2740,S
SINM	REAL*8		50	1030,D 2800,S
SIVFEQ	REAL*8			940,D
T	REAL*8		50	1020,D 2690,S 3030,U
TAPES	REAL*8			1110,D
TEORCE	REAL*8			4520,U
TH	REAL*8		50,5,2	1080,D 3300,S 3630,S 3640,S 3770,S 3770,U 3890,S 3890,U 3910,S 4460,S 4580,U 4770,S 4770,U 4030,U
THCDE	REAL*8			4510,U
THCON	REAL*8			1000,D
THER	REAL*8			1080,D
THETA	REAL*8			1040,D 1340,D 1810,S 1940,U 1940,S
THETAS	REAL*8			1040,U
THI	REAL*8		5	1130,D 4420,S 4460,U
TIM	REAL*8			3900,S 4050,U
TIME	REAL*8			1010,D 4910,U 5010,U
TIMED	REAL*8			1070,D 3280,S 3500,U
TMET	REAL*8			1010,D
TOTIME	REAL*8			1010,D 1320,S 1780,U 3800,U 4000,U
TPRNT	REAL*8			3500,S 3510,U 4050,S 4060,U 5010,S 5020,U

INPUT

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
TR140P	REAL*8			3210,U
TD	REAL*8			1010,D 3280,S 3790,S 3960,S 4910,U
TJ	REAL*8			1010,D 3280,S 3790,U 3800,S 3960,U 3980,S 3990,U 4000,S 4910,U
XKFFP	REAL*8			3380,S 3400,U 3410,S 3430,U
YN	REAL*8		6550	940,D 1150,S 1150,S 3130,S 3170,U
XP	REAL*8		6550	1000,D 3140,S 3190,U
Z	REAL*8		51	1120,D 2770,S

MAIN

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED												
CARD	REAL*8		20	130,D	140,S	300,S	310,U	330,U	330,U	390,U	400,U					
CONST	REAL*8			50,D												
CYCLE	REAL*8			110,D												
DELTE	REAL*8			70,D	580,U	580,U	530,U	730,U								
DELETED	REAL*8			100,D												
DT2	REAL*8			50,D												
FORCE	REAL*8		2040	30,D												
HARM	REAL*8			90,D												
I	INTEGER			510,D	520,U											
IDCDE	INTEGER			60,D												
IDELE	INTEGER			60,D												
IHARM	INTEGER		5	90,D												
INDIUT	INTEGER			470,U	670,D											
IPRINT	INTEGER			80,D												
IRSTRY	INTEGER			100,D	500,U	610,U										
ITAM	INTEGER			110,D	620,S	630,U	650,U	690,P	720,S	740,P						
ITP	INTEGER			100,D	550,S	580,U	610,U	630,U								
KKP2	INTEGER			580,S	500,U	620,U	650,U	720,U								
LARGE	INTEGER			490,S	690,P	700,U										
LL	INTEGER			80,D	600,S	610,S	620,U									
MPRINT	INTEGER			290,S	340,U	370,S	380,S	380,U								
N	INTEGER			50,D												
NCARD	INTEGER			270,S	310,U	320,S	320,U	410,U								
NCASE	INTEGER			220,S 750,U	280,S	280,U	350,U	420,S	420,U	430,U	460,S	480,S	480,U			
NCASES	INTEGER			200,S	210,U	430,U	750,U									
ND	INTEGER			120,D	200,S	260,U	330,U	440,U								

MAIN

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
NFLMS	INTEGER			50,0
NFO	INTEGER			50,0
NEQT	INTEGER			50,0 510,0
NH	INTEGER			50,0
NHNS	INTEGER			50,0
NHP	INTEGER			90,0
NLTERM	INTEGER			740,0
NN	INTEGER			50,0
NNODES	INTEGER			50,0
NOIT	INTEGER			80,0 590,5
NPRNT	INTEGER			100,0
NPRNT	INTEGER			100,0
NPRNTE	INTEGER			60,0
NPRNTE	INTEGER			60,0
NPRNTO	INTEGER			50,0
NS	INTEGER			120,0 200,5
NSIZE	INTEGER			50,0
NT	INTEGER			120,0
PRINT	REAL*8			80,0
QV	REAL*8		1020	30,0 140,5
QV1	REAL*8		1020	30,0
QV2	REAL*8		1020	30,0
QP	REAL*8		1020	30,0
QP1	REAL*8		1020	30,0 520,5
QS	REAL*8			30,0
RESTRT	REAL*8			100,0

MAIN

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED						
SETUP	REAL*8			690,U						
TAPES	REAL*8			120,D						
TFST	REAL*8	END		150,S	310,U	330,U	400,U			
TIME	REAL*8			70,D	630,S	640,U	690,P	730,S	730,U	
TIMED	REAL*8			100,D	560,S	580,U	630,U			
TMET	REAL*8			70,D						
TOTIME	REAL*8			70,D	580,U					
T0	REAL*8			70,D	570,S	660,S				
T1	REAL*8			70,D	640,U	660,U				

MATMUT

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
A	REAL*8		204	10850,U 10870,D 10950,S 11040,S 11040,S 11050,S 11050,U 11060,S 11060,U 11070,S 11070,U 11190,U 11210,U
FORCE	REAL*8		1020	10850,U 10870,D 11040,U 11050,U 11060,U 11060,U 11070,U 11070,U 11080,U 11180,U
I	INTEGER			10940,S 10950,U 11020,S 11040,U 11040,U 11110,S 11120,U
IH	INTEGER			10850,U 10980,U 10990,U
IID	INTEGER			10850,U 10940,U 10960,U 10990,U
J	INTEGER			11010,S 11020,U 11040,U 11140,U 11150,U 11180,U 11190,U 11190,U 11210,U 11210,U
JM1	INTEGER			11150,S 11160,U
L	INTEGER			11160,S 11180,U 11180,U 11190,U
MATMUT	INTEGER			10850,D
NA	INTEGER			11030,S 11030,S 11030,S 11040,U
NE	INTEGER			10990,S 11040,U 11050,U 11060,U 11060,U 11070,U 11070,U 11080,U 11130,U
NEE	INTEGER			11130,S 11180,U 11190,U 11210,U
NI	INTEGER			11120,S 11130,U 11180,U 11180,U 11190,U 11190,U 11210,U 11210,U
NK	INTEGER			10980,S 11000,U 11050,U 11060,U 11060,U 11070,U 11070,U 11070,U 11090,U
NKK	INTEGER			11090,S 11170,S 11170,U 11180,U 11190,U 11200,S 11200,S 11210,U
NM1	INTEGER			11100,S 11110,U
NN	INTEGER			10960,S 10970,U 11100,U
NT	INTEGER			10970,S 10980,U
STEM	REAL*8		6550	10850,U 10870,D 11040,U 11050,U 11060,U 11060,U 11070,U 11070,U 11070,U 11180,U 11190,U 11210,U

## NLTFPM

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
ARCL	REAL*8		50	6610,D
CP1	REAL*8			6640,D 6780,S
CP2	REAL*8			6640,D 6790,S
CONST	REAL*8			6550,D
COSINE	REAL*8		51	6610,D
COSM	REAL*8		50	6610,D
DD1	REAL*8			6640,D 6820,S
DD2	REAL*8			6640,D 6830,S
DELTA	REAL*8			6570,D
DT2	REAL*8			6550,D
EFS	REAL*8			6620,D
ES	REAL*8		5	6620,D
EFT	REAL*8		5	6620,D
ET	REAL*8		5	6620,D
E1	REAL*8		50	6620,D 6780,U 6820,U
E13	REAL*8		5	6620,D
E2	REAL*8		50	6620,D 6790,U 6830,U
E23	REAL*8		5	6620,D
EV	REAL*8			6770,S 6780,U 6790,U 6820,U 6830,U
ENU1	REAL*8		50	6620,D 6770,U
ENU2	REAL*8		50	6620,D 6770,U
FORCE	REAL*8		2040	6580,D
G	REAL*8		50	6600,D 6800,U 6810,U
GG0	REAL*8			6640,D
GEOM	REAL*8			6620,D
GG1	REAL*8			6640,D 6800,S



NLTERM

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
GG2	REAL*8			6640,D 6810,S
I	INTEGER			6730,S 6740,U
IDCOF	INTEGER			6560,D
IDEIF	INTEGER			6560,D
II	INTEGER			6950,S 6960,U 6990,U 7000,U
IPRINT	INTEGER			6660,D
ITAM	INTEGER			6530,U 6920,U 6920,U 6920,U 6920,U 6930,P
II	INTEGER			6760,S 6770,U 6770,U 6780,U 6780,U 6790,U 6790,U 6800,U 6800,U 6810,U 5810,U 6820,U 6820,U 6830,U 6830,U 6870,P 6930,P 6960,U
JJ	INTEGER			6970,S 6980,U 6990,U 7000,U
KA	INTEGER			6960,S 6980,U
KK	INTEGER			6980,S 6990,U 6990,U 7000,U
LL	INTEGER			6640,D
N	INTEGER			6550,D
NCLOST	INTEGER			6650,D 6930,U
NFLEMS	INTEGER			6550,D 6760,U
NFQ	INTEGER			6550,D 6960,U
NFQT	INTEGER			6550,D 6730,U
NH	INTEGER			6550,D 6950,U
NHNS	INTEGER			6550,D
NLTERM	INTEGER			6530,D
NLTRMS	INTEGER			6620,D
NN	INTEGER			6550,D
NNODES	INTEGER			6550,D
NOIT	INTEGER			6660,D 6930,U
NPRINT	INTEGER			6560,D
NDONTL	INTEGER			6560,D

## NLTERM

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
NPRINT	INTEGER			6550,D
NCTZF	INTEGER			6550,D
NSTRSS	INTEGER			6650,D 6910,U 6920,U 6920,U 6930,U
NTHETA	INTEGER			6650,D
PH	REAL*8		50	6610,D
PHD	REAL*8		50	6610,D
PRINT	REAL*8			6660,D
QV	REAL*8		1020	6580,D
QV1	REAL*8		1020	6580,D
QV2	REAL*8		1020	6580,D
QP	REAL*8		1020	6590,D 6740,S 6990,S 6990,U 7000,S
QPR	REAL*8		8,5	6630,D 6990,U 7000,U
QPRIME	REAL*8			6870,U
QPI	REAL*8		1020	6580,D
QS	REAL*8			6580,D
R	REAL*8		50	6610,D
SINF	REAL*8		51	6600,D
SINM	REAL*8		50	6610,D
STRESS	REAL*8			6930,U
T	REAL*8		50	6600,D 6780,U 6790,U 6800,U 6910,U 6820,U 6830,U
THETA	REAL*8		20	6650,D
THETAS	REAL*8			6650,D
TIME	REAL*8			6570,D
TMFT	REAL*8			6570,D
TOTIME	REAL*8			6570,D
T0	REAL*8			6570,D

NLTERM

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
T1	REAL*8			6570.0

NRESTP

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
CONST	REAL*8			10280,D
DT2	REAL*8			10280,D
FORCE	REAL*8		204	10270,D 10720,S
I	INTEGER			10420,S 10440,U 10440,U 10440,U 10470,S 10480,U 10480,U 10480,U 10540,S 10550,U 10570,U 10570,U 10570,U 10570,U 10610,S 10620,U 10620,U 10620,U 10620,U 10690,S 10700,U 10700,U 10700,U 10700,U 10800,S 10810,U
IDCDE	INTEGER			10290,D
IOELF	INTEGER			10290,D
IPLM	INTEGER			10400,S 10410,U 10540,U 10570,U 10590,U 10600,U 10620,U 10640,U 10660,U 10670,S 10670,U 10700,U
J	INTEGER			10430,S 10440,U 10460,S 10480,U 10560,S 10570,U 10590,S 10620,U 10640,U 10680,S 10680,U 10700,U
K	INTEGER			10440,S 10450,U 10480,S 10490,U 10500,S 10510,U 10570,S 10580,U 10620,S 10630,U 10640,S 10650,U 10700,S 10710,U 10810,S 10820,U
KY	INTEGER			10250,U 10360,U 10780,U
L	INTEGER			10380,S 10390,U
LK	INTEGER		204	10300,D 10390,U
M	INTEGER			10280,D
N	INTEGER			10550,S 10560,U 10600,S 10610,U 10640,U 10640,U 10640,U 10640,U 10640,U
NCLOSE	INTEGER			10300,D 10350,U
NFLEMS	INTEGER			10280,D 10660,U
NFQ	INTEGER			10300,U 10400,U 10420,U 10430,U 10460,U 10470,U 10500,U 10500,U 10500,U 10500,U 10540,U 10590,U 10600,U 10720,U
NFQT	INTEGER			10280,D
NH	INTEGER			10280,D
NHNS	INTEGER			10280,D
NIX	INTEGER			10280,D
NV	INTEGER			10280,D 10440,U 10480,U 10500,U 10570,U 10620,U 10640,U 10700,U
VNODES	INTEGER			10280,D 10800,U

NPESTR

VARIABLE	TYPE	INITIAL VAL IF	DIMENSION	WHERE/HOW USED
NDDRE	INTEGER			10350,S 10360,S 10360,U 10370,U 10380,U
NDDRES	INTEGER			10390,D 10350,U
NPRNTE	INTEGER			10200,D
NPRNTL	INTEGER			10200,D
NPRNTQ	INTEGER			10200,D
NPESTR	INTEGER			10250,D
NSIZE	INTEGER			10280,D
RSTRNT	REAL#R			10370,D
SIVEFO	REAL#B			10270,D
STEM	REAL#R		6550	10270,D 10450,S 10490,S 10510,S 10580,S 10630,S 10650,S 10710,S 10790,S 10820,S

OPTIME

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
ALS	REAL*8		50	7170,D 8820,U 8830,U
ALT	REAL*8		50	7170,D 8820,U 8830,U
APCL	REAL*8		50	7140,D
ARCLY	REAL*8			7380,S 7440,U 7450,U 7460,U 7470,U 7480,U 7490,U
ARL	REAL*8			7330,S 7340,U 7350,U 7380,U 7900,U 8670,U
CCC	REAL*8		125	7090,D 8100,U 8120,U 9150,U 8820,U
CCCC	REAL*8		625	7100,D 8520,U
CC1	REAL*8			7150,D 8100,U 8120,U 8150,U 8170,U 8510,U 8520,U 8820,U
CC2	REAL*8			7150,D 8120,U 8170,U 8540,U 8830,U
CES	REAL*8			7920,S 8100,S 8100,U 8210,U 8240,U 8260,U 8290,U
CFST	REAL*8			7940,S 8140,S 8140,U 8210,U 8230,U 8240,U 8260,U 8280,U 8290,U
CET	REAL*8			7930,S 8120,S 8130,U 8230,U 8240,U 8280,U 8290,U
CF13	REAL*8			7950,S 8150,S 8160,U 8210,U 8240,U 8260,U 8290,U 9690,S 8820,S 8820,U 8860,U 8880,U 8890,U 8910,U
CF23	REAL*8			7960,S 8170,S 8180,U 8210,U 8230,U 8240,U 8260,U 8280,U 8290,U 8700,S 8820,S 8830,U 8860,U 8870,U 8880,U 8890,U 8900,U 8910,U
CF413	REAL*8			8340,S 8520,S 8520,S 8580,U 8600,U 8610,U 8630,U
CF423	REAL*8			8350,S 8540,S 8540,U 8580,U 8590,U 8600,U 8610,U 8620,U 8630,U
CL2R	REAL*8			7570,S 7650,U
CONST	REAL*8			7070,D
COPIH	REAL*8			7350,S 7420,U 7450,U 7470,U 7550,U 7570,U
COSINE	REAL*8		51	7140,D
COSM	REAL*8		50	7140,D
CO2R	REAL*8			7550,S 7630,U
CS	REAL*8			7090,D
CSS	REAL*8		125	7090,D 8140,U 8170,U 8830,U
CS4	REAL*8			7100,D

QPRIME

VARIABLE	TYPE	INITIAL VAL JF	DIMENSION	WHERE/HOW USED			
DD1	REAL*8			7150,D			
DD2	REAL*8			7150,D			
DELTF	REAL*8			7190,D			
DRD	REAL*8			7310,S	7330,U	7330,U	7340,U
DSQRT	REAL*8			7330,U			
DTH	REAL*8		50,5,2	7170,D			
D*2	REAL*8			7070,D			
D7	REAL*8			7320,S	7330,U	7330,U	7350,U
EFS	REAL*8			7110,D			
ES	REAL*8		5	7110,D	7820,S	8150,U	8170,U
ESQ1	REAL*8			7500,S	7820,U	8210,U	
ESQ3	REAL*8			7510,S	7820,U	8240,U	
ESQ5	REAL*8			7520,S	7820,U	8260,U	
ESQ7	REAL*8			7530,S	7820,U	8290,U	
EST	REAL*8		5	7110,D	7830,S	8160,U	8180,U
ESTQ1	REAL*8		5	7230,D	7660,S	7830,U	8210,U
ESTQ2	REAL*8			7480,S	7830,U	8230,U	
ESTQ3	REAL*8		5	7230,D	7670,S	7830,U	8240,U
ESTQ5	REAL*8		5	7240,D	7690,S	7830,U	8260,U
ESTQ6	REAL*8			7490,S	7840,U	8290,U	
ESTQ7	REAL*8		5	7240,D	7600,S	7840,U	8290,U
ET	REAL*8		5	7110,D	7800,S	8150,U	8170,U
ETQ2	REAL*8		5	7240,D	7700,S	7710,U	7800,U 8230,U
ETQ3	REAL*8			7400,S	7800,U	8240,U	
ETQ6	REAL*8		5	7240,D	7710,S	7800,U	8280,U
ETQ7	REAL*8			7410,S	7800,U	8200,U	

OPTIME

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
E1	REAL*8		50	7130,D
F13	REAL*8		5	7110,D 7850,S 8100,U 8120,U 8120,U 8120,U 8140,U 8150,U 8170,U 8510,U 8520,U 8520,U 8520,U 8540,U 8820,U
F1301	REAL*8			7440,S 7510,U 7850,U 8210,U 8580,U 8860,U
F1303	REAL*8			7450,S 7500,U 7850,U 8240,U 8600,U 8880,U
F1305	REAL*8			7460,S 7530,U 7850,U 8260,U 8610,U 8890,U
F1307	REAL*8			7470,S 7520,U 7850,U 8290,U 8630,U 8910,U
F2	REAL*8		50	7130,D
F23	REAL*8		5	7110,D 7860,S 8100,U 8100,U 8120,U 8120,U 8140,U 8160,U 8170,U 8510,U 8520,U 8540,U 8540,U 8540,U 8830,U
F2301	REAL*8		5	7230,D 7620,S 7670,U 7860,U 8210,U 8580,U 8860,U
F2302	REAL*8			7420,S 7430,U 7860,U 8230,U 8590,U 8870,U
F2303	REAL*8		5	7230,D 7630,S 7660,U 7860,U 8240,U 8600,U 8880,U
F2305	REAL*8		5	7230,D 7640,S 7690,U 7860,U 8260,U 8610,U 8890,U
F2306	REAL*8			7430,S 7870,U 8280,U 8620,U 8900,U
F2307	REAL*8		5	7230,D 7650,S 7680,U 7870,U 8290,U 8630,U 8910,U
FNU1	REAL*8		50	7130,D 8100,U 8120,U 8150,U 8170,U 8510,U 8820,U
FNU2	REAL*8		50	7130,D 8830,U
FDR	REAL*8			8510,S 8520,U 8540,U
FORCE	REAL*8		2040	7200,D
G	REAL*8		50	7130,D
GCD	REAL*8			7150,D
GEOM	REAL*8			7130,D
GG1	REAL*8			7150,D 8140,U 8150,U 8170,U 8510,U
GG2	REAL*8			7150,D
HARM	REAL*8			7160,D



QBRTMR

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
I	INTEGER			7980,S 7990,U 8010,S 8020,U 8030,S 8040,U 8050,U 8060,U 8070,U 8080,U 8090,U 8100,U 8110,U 8120,U 8130,U 8140,U 8150,U 8150,U 8170,U 8180,U 8190,U 8200,S 8210,U 8220,U 8230,U 8240,U 8250,U 8260,U 8270,S 8280,U 8290,U 8300,U
IABS	INTEGER			8070,U 8420,U 8460,U 8780,U
ITCOF	INTEGER			7080,D
ITELF	INTEGER			7080,D
ITD	INTEGER			8320,S 8400,S 8400,U 8520,U 8520,U 8540,U 8540,U
ITM	INTEGER			7500,S 7600,U 7620,U 7630,U 7640,U 7650,U 7660,U 7660,U 7660,U 7670,U 7670,U 7670,U 7680,U 7690,U 7690,U 7500,U 7700,U 7710,U 7710,U 7730,U 7800,U 7800,U 7800,U 7800,U 7820,U 7820,U 7830,U 7830,U 7830,U 7830,U 7840,U 7850,U 7860,U 7860,U 7860,U 7870,U 7870,U
IHARM	INTEGER		5	7160,D 7600,U 7970,U 8020,U 8040,U 8360,U 8380,U 8400,U 8440,U 8710,U 8730,U 8750,U
II	INTEGER			8020,S 8050,U 8060,U 8380,S 8410,U 8420,U 8730,S 8760,U 8770,U
IMJ	INTEGER			8060,S 8070,U 8420,S 8470,U 8470,U 8770,S 8780,U
IPJ	INTEGER			8050,S 8070,U 8410,S 8470,U 8470,U 8760,S 8780,U
ITCOF	INTEGER			7180,D
ITELF	INTEGER			7180,D 8650,U
ITM	INTEGER			7890,S 8080,S 8080,U 8100,U 8100,U 8120,U 8120,U 8140,U 8150,U 8150,U 8150,U 8170,U 8660,S 8800,S 8800,U 8820,U 8830,U
II	INTEGER			7050,U 7290,U 7320,U 7730,U 8100,U 8120,U 8150,U 8170,U 8510,U 8790,U 8700,U 8700,U 8820,U 8820,U 8820,U 8830,U 8830,U
J	INTEGER			8030,S 8040,U 8100,U 8100,U 8120,U 8120,U 8140,U 8150,U 8150,U 8160,U 8160,U 8170,U 8170,U 8170,U 8300,S 8400,U 8520,U 8520,U 8540,U 8540,U 8740,S 8750,U 8790,U 8790,U
JJ	INTEGER			8040,S 8050,U 8060,U 8400,S 8410,U 8420,U 8750,S 8760,U 8770,U
J1	INTEGER			7290,S 7310,U 7320,U 7360,U
J11	INTEGER			7300,U 7310,U 7320,U 7360,U
K	INTEGER			7600,S 7610,U 7970,S 8070,U 8070,U 8430,S 8440,U 8510,U 8520,U 8540,U 8710,S 8780,U 8780,U
KK	INTEGER			7730,S 7740,U 7750,U 7760,U 7770,U 7780,U 7790,U 8440,S 8450,U 8460,U

QPPIME

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
KK1	INTEGER			7740,U 7820,U 7830,U 7850,U 7860,U
KK2	INTEGER			7750,S 7800,U 7830,U 7860,U
KK3	INTEGER			7760,S 7800,U 7820,U 7830,U 7850,U 7860,U
KK5	INTEGER			7770,S 7820,U 7840,U 7850,U 7870,U
KK6	INTEGER			7780,S 7800,U 7840,U 7870,U
KK7	INTEGER			7790,S 7800,U 7820,U 7840,U 7850,U 7870,U
KML	INTEGER			8460,S 8470,U 8470,U
KPI	INTEGER			8450,S 8470,U 8470,U
L	INTEGER			8330,S 8360,U 8580,U 8580,U 8580,U 8590,U 8590,U 8600,U 8600,U 8600,U 8610,U 8610,U 8610,U 8620,U 8620,U 8630,U 8630,U 8630,U
LL	INTEGER			8360,S 8450,U 8460,U
M	INTEGER			7910,S 7970,U 7990,U 8210,U 8210,U 8210,U 8220,U 8230,U 8230,U 8230,U 8240,U 8240,U 8240,U 8250,U 8260,U 8260,U 8260,U 8270,U 8280,U 8280,U 8290,U 8290,U 8290,U 8290,U 8300,U 8680,U 8710,U 8860,U 8860,U 8860,U 8870,U 8870,U 8880,U 8880,U 8890,U 8890,U 8890,U 8900,U 8900,U 8900,U 8910,U 8910,U 8910,U
N	INTEGER			7070,D
NELEMS	INTEGER			7070,D
NFO	INTEGER			7070,D 7730,U
NEQT	INTEGER			7070,D
NH	INTEGER			7070,D 7590,U 7910,U 8010,U 8030,U 8330,U 8370,U 8390,U 8430,U 8680,U 8720,U 8740,U
NHNS	INTEGER			7070,D
NHP	INTEGER			7160,D
NLTRMS	INTEGER			7120,D
NN	INTEGER			7070,D
NNODES	INTEGER			7070,D
NPRINTF	INTEGER			7080,D
NPRINTH	INTEGER			7180,D



QPRIME

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
STPH	REAL*8			7340,S 7440,U 7460,U 7480,U 7490,U 7560,U 7580,U
SL2R	REAL*8			7560,S 7640,U
SD2R	REAL*8			7580,S 7620,U
SSC	REAL*8		125	7090,D 8100,U 8120,U 8150,U
SSCC	REAL*8		625	7100,D 8520,U
SSSS	REAL*8		625	7100,D 8540,U
T	REAL*8		50	7130,D
TH	REAL*8		50,5,2	7170,D 8790,U 8790,U 8790,U
THCON	REAL*8			7180,D
THFR	REAL*8			7170,D
THT	REAL*8			8790,S 8820,U 8830,U
TIME	REAL*8			7190,D 8790,U
TMET	REAL*8			7190,D
TOTIME	REAL*8			7190,D
T2	REAL*8			7190,D 8790,U 8790,U
T1	REAL*8			7190,D 8790,U
XX	REAL*8			7610,S 7620,U 7630,U 7640,U 7650,U 7700,U
Z	REAL*8		51	7220,D 7320,U 7320,U

SETUP

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
CC1	REAL*8			5730,D
CC2	REAL*8			5730,D
CONST	REAL*8			5710,D
CS	REAL*8			6120,S 6180,U
DABS	REAL*8			6330,U
DCOS	REAL*8			6120,U
DD1	REAL*8			5730,D
DD2	REAL*8			5730,D
DELTEP	REAL*8			5770,D
DSTN	REAL*8			6110,U
DY2	REAL*8			5710,D
FORCE	REAL*8		2040	5600,D
GCD	REAL*8			5730,D
GG1	REAL*8			5730,D
GG2	REAL*8			5730,D
HARM	REAL*8			5760,D
HOURQ1	REAL*8			5880,U
HOURQN	REAL*8			5870,U
I	INTEGER			5960,S 5970,U 5980,U 6060,S 6070,U 6250,S 6260,U 6270,U
IDCDE	INTEGER			5720,D
IDELE	INTEGER			5720,D
IH	INTEGER			5830,S 5840,U 5850,U 5850,U 5870,P 5880,P 6000,U
IHARM	INTEGER		5	5760,D 5840,U 6100,U
INDIT	INTEGER			6400,U
IPRINT	INTEGER			5750,D 5800,U 5800,U
IRSTRY	INTEGER			5770,D

## SF\*UP

VARIABLE	TYPE	INITIAL VALJF	DIMENSION	WHERE/HOW USED
IT	INTEGER			5890,S 5910,U 6050,S 6110,U 6120,U 6230,U
ITAM	INTEGER			5660,U 5820,U 5820,P 5870,U 5880,U 5890,U 5890,U 5890,U 5900,U 5910,U 5940,U 6220,U 6350,U 6390,U 6390,U
ITP	INTEGER			5770,D
ITJ	INTEGER			6390,S 6400,U
II	INTEGER			6130,S 6140,U
J	INTEGER			5980,U 5980,S 6160,S 6170,U 6180,U 6180,U 6180,U 6180,U 6200,U 6200,U 6200,U 6270,U 6270,U
JH	INTEGER			6080,S 6090,S 6100,U
K	INTEGER			5970,S 5980,U 6140,S 6150,U 6180,U 6180,U 6200,U 6200,U 6260,S 6270,U
KY	INTEGER			5860,S 5870,P 5880,P 5950,U
LARGE	INTEGER			5660,U 6340,S
LF	INTEGER			6330,U
LL	INTEGER			5750,D 5820,U 5870,U 5880,U
N	INTEGER			5710,D 5860,S 5970,U 6090,S 6150,U 6330,U
NFCOST	INTEGER			5740,D
NFLEMS	INTEGER			5710,D
NFO	INTEGER			5710,D 5860,U 6060,U 6090,U
NFQT	INTEGER			5710,D
NH	INTEGER			5710,D 5830,U 6000,U 6010,U 6080,U
NHNS	INTEGER			5710,D
NHP	INTEGER			5760,D
NLTERM	INTEGER			5820,U
NN	INTEGER			5710,D 5850,S
NNODES	INTEGER			5710,D 5960,U 6130,U 6250,U
NQIT	INTEGER			5750,D 5900,U
NDK	INTEGER			6150,S 6180,U 6200,U

SETUP

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
NPRNT	INTEGER			5770,D 6390,U 6390,U
NPRNT	INTEGER			5770,D 6390,U 6400,U
NPRNTE	INTEGER			5720,D
NPRNTE	INTEGER			5720,D
NPRNTQ	INTEGER			5710,D 5920,U
NSIZE	INTEGER			5710,D 5850,U
NSTRSS	INTEGER			5740,D
NTHETA	INTEGER			5740,D 6050,U
PRINT	REAL*8			5750,D
QLOAD	REAL*8		204	5680,D 6070,S 6180,S 6180,U 6200,S 6200,U 6270,U
QN	REAL*8		1020	5600,D 5980,U 6180,U 6200,U 6330,U
QN1	REAL*8		1020	5600,D
QN2	REAL*8		1020	5600,D
QP	REAL*8		1020	5600,D
QP1	REAL*8		1020	5600,D
QS	REAL*8			5690,D
RESTRT	REAL*8			5770,D
SETUP	REAL*8			5660,D
SLVFFD	REAL*8			5680,D
SN	REAL*8			6110,S 6200,U
THETA	REAL*8		20	5740,D 6110,U 6120,U 6230,U
THETAS	REAL*8			5740,D
THETAJ	REAL*8			6230,S 6240,U
TIME	REAL*8			5660,U 5930,U 6350,U
TIMFP	REAL*8			5770,D
TPRNT	REAL*8			5930,S 5940,U 6220,U

SETUP

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
XIH1	REAL#8			6100,S 6110,U 6120,U
XV	REAL#8		6550	5680,D



SOLVED

VARIABLE	TYPE	INITIAL VAL/JF	DIMENSION	WHERE/HOW USED
A	REAL*8		1310	11310,D 11410,S 11460,S 11460,U 11460,U 11470,S 11470,U 11470,U 11470,U 11470,U 11470,U 11470,U 11470,U 11480,S 11480,U 11480,U 11480,U 11480,S 11480,U 11480,U 11480,U 11480,U 11480,U 11480,U 11480,U 11480,U 11500,U 11510,S 11510,U 11510,U 11520,S 11520,U 11520,U 11520,U 11520,U 11520,U 11520,U 11520,U 11520,U 11530,U 11530,U 11530,S 11530,U 11530,U 11530,U 11530,U 11530,U 11530,U 11530,U 11530,U 11530,U 11530,U 11540,U 11550,S 11550,U 11550,U 11550,U 11560,S 11560,U 11560,U 11560,U 11560,U 11560,U 11560,U 11560,U 11570,U 11570,U 11570,U 11570,U 11570,U 11570,U 11570,U 11570,U 11570,U 11570,U 11570,U 11570,U 11570,U 11580,U 11600,U 11600,U 11600,U 11600,U 11650,U 11660,U 11670,U 11680,U 11690,U 11700,U 11700,U 11700,U 11720,U 11730,U 11740,U 11750,U 11760,U 11770,U 11780,U 11780,U 11790,U 11800,U 11810,U 11810,U 11810,U 11820,U 11830,U 11840,U 11850,U 11860,U 11870,U 11880,U 11880,U 11890,U 11900,U 11900,U 11900,U 11900,U 11920,U 11930,U 11940,U 11950,U 11960,U 11970,U 11980,U 11980,U 11990,U 12000,U 12000,U 12000,U 12000,U 12020,U 12030,U 12040,U 12050,U 12060,U 12070,U 12080,U 12080,U 12090,U 12100,U 12100,U 12100,U 12100,U 12640,S 12650,S 12680,S 12690,S 12700,S 12710,S 12720,S 12720,S 12730,S 12740,S 12740,S 12740,S 12740,S 12760,S 12770,S 12780,S 12790,S 12800,S 12810,S 12820,S 12820,S 12830,S 12840,S 12840,S 12840,S 12840,S 12860,S 12870,S 12880,S 12890,S 12900,S 12910,S 12920,S 12920,S 12930,S 12940,S 12940,S 12940,S 12940,S 12960,S 12970,S 12980,S 12990,S 13000,S 13010,S 13020,S 13030,S 13040,S 13050,S 13060,U 13070,U 13070,U 13070,U 13080,U 13080,U 13080,U 13090,U 13100,U 13100,U 13100,U 13100,U 13100,U 13100,U 13100,U 13100,U 13120,U 13130,U 13140,U 13150,U 13150,U 13150,U 13160,U 13160,U 13160,U 13160,U 13160,U 13160,U 13160,U 13170,U 13170,U 13180,U 13180,U 13180,U 13180,U 13180,U 13180,U 13240,U 13250,U 13250,U 13250,U 13250,U 13260,U 13260,U 13270,U 13270,U 13270,U 13270,U 13270,U 13270,U 13320,U 13320,U 13320,U 13320,U 13320,U 13330,U 13340,U 13340,U 13340,U 13340,U 13350,U 13350,U 13350,U 13360,U 13360,U 13360,U 13360,U 13360,U 13360,U 13370,U 13370,U 13370,U 13380,U 13380,U 13380,U 13380,U 13380,U 13380,U 13380,U 13380,U 13380,U 13390,U 13390,U 13410,U 13410,U 13410,U 13410,U 13410,U 13420,U 13430,U 13430,U 13430,U 13430,U 13430,U 13430,U 13440,U 13440,U 13450,U 13450,U 13450,U 13450,U 13450,U 13460,U 13460,U 13460,U 13460,U 13460,U 13470,U 13470,U 13470,U 13470,U 13480,U 13480,U 13480,U 13550,U
AM1	REAL*8			11910,U 12230,U 12310,U 12330,U 12400,U 12430,U 12510,U 12550,U 12770,U
AM10	REAL*8			11650,S 11720,S 12200,U 12230,U 12280,U 12330,U 12370,U 12430,U 12480,U 12550,U 12600,U 12680,U
AM2	REAL*8			11800,S 12160,U 12170,U 12180,U 12190,U 12760,U
AM3	REAL*8			11790,S 12160,U 12170,U 12180,U 12190,U 12750,U
AM4	REAL*8			11780,S 12160,U 12170,U 12180,U 12190,U 12740,U
AM5	REAL*8			11700,S 11770,S 12220,U 12230,U 12300,U 12330,U 12390,U 12430,U 12500,U 12550,U 12650,U 12730,U
AM6	REAL*8			11690,S 11760,S 12120,U 12130,U 12140,U 12150,U 12640,U 12720,U
AM7	REAL*8			11680,S 11750,S 12120,U 12130,U 12140,U 12150,U 12630,U 12710,U
AM8	REAL*8			11670,S 11740,S 12210,U 12230,U 12290,U 12330,U 12380,U 12430,U 12490,U 12550,U 12620,U 12700,U
AM9	REAL*8			11660,S 11730,S 12080,U 12090,U 12100,U 12110,U 12610,U 12690,U
A7	REAL*8			11820,S 12090,U 12120,U 12160,U 12200,S 12200,U 12230,U 12230,U 12240,U 12250,U 12260,U 12270,U 12780,U

SOLVED

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
A1	REAL*8			11830,S 12080,S 12080,U 12120,U 12160,U 12210,S 12210,S 12230,U 12230,U 12240,U 12250,U 12260,U 12270,U 12700,U
A10	REAL*8			11920,S 12330,S 12330,U 12420,U 12440,U 12530,U 12560,U 12880,U
A11	REAL*8			11930,S 12100,U 12140,U 12180,U 12250,U 12340,U 12370,S 12370,U 12430,U 12430,U 12450,U 12460,U 12800,U
A12	REAL*8			11940,S 12100,S 12100,U 12140,U 12180,U 12250,U 12340,U 12380,S 12380,U 12430,U 12430,U 12450,U 12460,U 12900,U
A13	REAL*8			11950,S 12140,S 12140,U 12180,U 12250,U 12340,U 12390,S 12390,U 12430,U 12430,U 12450,U 12460,U 12910,U
A14	REAL*8			11960,S 12180,S 12180,U 12250,U 12340,U 12400,S 12400,U 12430,U 12430,U 12450,U 12460,U 12920,U
A15	REAL*8			11970,S 12250,S 12250,U 12340,U 12410,S 12410,U 12430,U 12430,U 12450,U 12460,U 12930,U
A16	REAL*8			11980,S 12340,S 12340,U 12420,S 12420,U 12440,U 12440,U 12450,U 12470,U 12940,U
A17	REAL*8			11990,S 12430,S 12430,U 12540,U 12560,U 12950,U
A18	REAL*8			12000,S 12110,U 12150,U 12190,U 12260,U 12350,U 12450,U 12480,S 12480,U 12550,U 12550,U 12570,U 12960,U
A19	REAL*8			12010,S 12110,S 12110,U 12150,U 12190,U 12260,U 12350,U 12450,U 12490,S 12490,U 12550,U 12550,U 12570,U 12970,U
A2	REAL*8			11840,S 12120,S 12120,U 12160,U 12220,S 12220,U 12230,U 12230,U 12240,U 12250,U 12260,U 12270,U 12800,U
A20	REAL*8			12020,S 12150,S 12150,U 12190,U 12260,U 12350,U 12450,U 12500,S 12500,U 12550,U 12550,U 12570,U 12980,U
A21	REAL*8			12030,S 12190,S 12190,U 12260,U 12350,U 12450,U 12510,S 12510,U 12550,U 12550,U 12570,U 12990,U
A22	REAL*8			12040,S 12260,S 12260,U 12350,U 12450,U 12520,S 12520,U 12550,U 12550,U 12570,U 13000,U
A23	REAL*8			12050,S 12350,S 12350,U 12450,U 12530,S 12530,U 12560,U 12560,U 12580,U 13010,U
A24	REAL*8			12060,S 12450,S 12450,U 12540,S 12540,U 12560,U 12560,U 12580,U 13020,U
A25	REAL*8			12070,S 12550,S 12550,U 13030,U
A3	REAL*8			11850,S 12160,S 12160,U 12230,U 12230,U 12240,U 12250,U 12260,U 12270,U 12810,U

SOLVED

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
A4	REAL*8			11860,S 12230,S 12230,U 12320,U 12330,U 12410,U 12430,U 12520,U 12550,U 12820,U
A5	REAL*8			11870,S 12090,U 12130,U 12170,U 12240,U 12280,S 12280,U 12330,U 12330,U 12340,U 12350,U 12360,U 12830,U
A6	REAL*8			11890,S 12090,S 12090,U 12130,U 12170,U 12240,U 12290,S 12290,U 12330,U 12330,U 12340,U 12350,U 12360,U 12840,U
A7	REAL*8			11890,S 12130,S 12130,U 12170,U 12240,U 12300,S 12300,U 12330,U 12330,U 12340,U 12350,U 12360,U 12850,U
A8	REAL*8			11900,S 12170,S 12170,U 12240,U 12310,S 12310,U 12330,U 12330,U 12340,U 12350,U 12360,U 12860,U
A9	REAL*8			11910,S 12240,S 12240,U 12320,S 12320,U 12330,U 12330,U 12340,U 12350,U 12360,U 12870,U
CONST	REAL*8			11270,D
CYCLE	REAL*8			11370,D
DT2	REAL*8			11270,D
I	INTEGER			11300,S 11400,U 11410,U 11420,S 11430,U 11430,U 11620,S 11650,U 11660,U 11670,U 11680,U 11690,U 11700,U 11720,U 11730,U 11740,U 11750,U 11760,U 11770,U 11780,U 11790,U 11800,U 11810,U 11820,U 11830,U 11840,U 11850,U 11860,U 11870,U 11880,U 11890,U 11900,U 11910,U 11920,U 11930,U 11940,U 11950,U 11960,U 11970,U 11980,U 11990,U 12000,U 12010,U 12020,U 12030,U 12040,U 12050,U 12060,U 12070,U 12080,U 12610,U 12620,U 12630,U 12640,U 12650,U 12680,U 12690,U 12700,U 12710,U 12720,U 12730,U 12740,U 12750,U 12760,U 12770,U 12780,U 12790,U 12800,U 12810,U 12820,U 12830,U 12840,U 12850,U 12860,U 12870,U 12880,U 12890,U 12900,U 12910,U 12920,U 12930,U 12940,U 12950,U 12960,U 12970,U 12980,U 12990,U 13000,U 13010,U 13020,U 13030,U 13500,S 13510,U 13510,U 13530,S 13540,U 13550,U
IDCOF	INTEGER			11280,D
IDEIF	INTEGER			11280,D
IH	INTEGER			11240,U
IPRINT	INTEGER			11290,D
ITAM	INTEGER			11320,D 11440,U 13520,U
J	INTEGER			11630,S 12270,U 12270,U 12270,U 12270,U 12270,U 12270,U 12270,U 12360,U 12360,U 12360,U 12360,U 12360,U 12460,U 12460,U 12460,U 12460,U 12460,U 12460,U 12460,U 12460,U 12460,U 12470,U 12570,U 12570,U 12570,U 12570,U 12570,U 12570,U 12570,U 12570,U 12580,U
K	INTEGER			11610,S 11620,U 11630,U 13090,S 13100,U 13110,U 13280,S 13290,U 13300,U

SOLVE0

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
KEY	INTEGER			11450,S 11640,U 12590,U 12660,S
L	INTEGER			13110,U 13120,U 13120,U 13120,U 13120,U 13130,S 13130,U 13130,U 13130,U 13130,U 13140,U 13150,U 13150,U 13150,U 13160,U 13160,U 13160,U 13170,U 13170,U 13170,U 13170,U 13180,U 13180,U 13180,U 13180,U 13200,S 13310,U 13320,U 13320,U 13420,U 13320,U 13330,U 13330,U 13340,U 13340,U 13340,U 13340,U 13350,U 13350,U 13360,U 13360,U 13360,U 13360,U 13360,U 13370,U 13370,U 13370,U 13380,U 13380,U 13380,U 13380,U 13390,U 13390,U 13390,U 13390,U 13410,U 13410,U 13410,U 13410,U 13410,U 13420,U 13430,U 13430,U 13430,U 13430,U 13440,U 13440,U 13450,U 13450,U 13450,U 13450,U 13460,U 13460,U 13460,U 13470,U 13470,U 13470,U 13470,U 13480,U 13480,U 13480,U 13480,U
LL	INTEGER			11290,S 11440,U 13520,U
M	INTEGER			13230,S 13240,U 13240,U 13250,U 13250,U 13250,U 13250,U 13260,U 13260,U 13260,U 13260,U 13270,U 13270,U 13270,U 13270,S 13270,U
N	INTEGER			13100,U 13120,U 13120,U 13120,U 13120,U 13120,U 13120,U 13120,U 13120,U 13130,U 13130,U 13130,U 13130,U 13130,U 13130,U 13140,U 13150,U 13150,U 13150,U 13150,U 13150,U 13150,U 13150,U 13160,U 13160,U 13170,U 13170,U 13170,U 13170,U 13170,U 13170,U 13170,U 13170,U 13180,U 13180,U 13180,U 13220,S 13240,U 13250,U 13250,U 13260,U 13260,U 13260,U 13270,U 13270,U 13270,U 13270,U 13270,U 13300,S 13320,U 13320,U 13320,U 13320,U 13320,U 13320,U 13330,U 13340,U 13340,U 13340,U 13340,U 13340,U 13350,U 13350,U 13360,U 13360,U 13360,U 13360,U 13360,U 13360,U 13370,U 13370,U 13370,U 13370,U 13380,U 13380,U 13380,U 13380,U 13380,U 13380,U 13390,U 13390,U 13390,U 13410,U 13410,U 13410,U 13410,U 13410,U 13420,U 13430,U 13430,U 13430,U 13430,U 13430,U 13430,U 13430,U 13440,U 13450,U 13450,U 13450,U 13450,U 13450,U 13450,U 13450,U 13460,U 13460,U 13460,U 13470,U 13470,U 13470,U 13470,U 13470,U 13470,U 13480,U 13480,U 13480,U
NELEMS	INTEGER			11270,U 11610,U 13090,U 13220,U 13230,U 13280,U 13290,U 13300,U
NEQ	INTEGER			11270,D 11420,U 13500,U
NEQT	INTEGER			11270,D
NH	INTEGER			11270,D
NHNS	INTEGER			11270,D
NI	INTEGER			11270,D 11400,U 13540,U
NNODES	INTEGER			11270,D
NNDI	INTEGER			11400,S 11410,U 13540,S 13550,S
NOIT	INTEGER			11290,D
NPDNTE	INTEGER			11280,D
NORNTL	INTEGER			11290,D
NPRNTO	INTEGER			11270,D

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SOLVED

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
NSIZE	INTEGER			11270,D 11390,U 13530,U
NW	INTEGER			11270,D
PRINT	REAL*R			11290,D
R	REAL*R		274	11310,D 11430,S 11580,S 11580,U 11580,U 11590,S 11590,U 11590,U 11590,U 11590,U 11600,S 11600,U 11600,U 11600,U 11600,U 12270,S 12270,U 12270,U 12270,U 12270,U 12270,U 12270,U 12360,S 12360,U 12360,U 12360,U 12360,U 12360,U 12360,U 12360,U 12460,S 12460,U 12460,U 12460,U 12460,U 12460,U 12460,U 12470,U 12570,S 12570,U 12570,U 12570,U 12570,U 12570,U 13070,U 13080,S 13080,U 13080,U 13080,U 13080,U 13080,U 13120,S 13120,U 13120,U 13120,U 13120,U 13120,U 13130,S 13130,U 13130,U 13130,U 13130,U 13130,U 13130,U 13140,U 13150,S 13150,U 13150,U 13150,U 13150,U 13150,U 13160,U 13160,U 13170,S 13170,U 13170,U 13170,U 13170,U 13170,U 13180,U 13180,U 13180,U 13240,S 13240,U 13250,S 13250,U 13250,U 13260,S 13260,U 13260,U 13260,U 13260,U 13270,S 13270,U 13270,U 13270,U 13320,S 13320,U 13320,U 13320,U 13320,U 13340,S 13340,U 13340,U 13340,U 13340,U 13340,U 13340,U 13350,U 13360,S 13360,U 13360,U 13360,U 13360,U 13360,U 13370,U 13370,U 13380,S 13380,U 13380,U 13380,U 13380,U 13380,U 13390,U 13390,U 13390,U 13390,U 13410,S 13410,U 13410,U 13410,U 13410,U 13410,U 13420,S 13430,U 13430,U 13430,U 13430,U 13430,U 13440,U 13450,S 13450,U 13450,U 13450,U 13450,U 13450,U 13460,U 13460,U 13470,S 13470,U 13470,U 13470,U 13470,U 13470,U 13470,U 13480,U 13480,U 13480,U 13510,U
SLVFFQ	REAL*R			11260,D
SOLVED	REAL*R			11240,D
SPA	REAL*R		6557	11260,D 11410,U 13550,S
SQR	REAL*R		274	11260,D 11430,U 13510,S

STRESS

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
AAV	REAL*8			14590,S 14600,U 14620,U
ALS	REAL*8		50	13710,D 14000,U 14150,U
ALT	REAL*8		50	13710,D 14100,U 14180,U
ARCI	REAL*8		50	13650,D 14140,U 14210,U 14210,U 14590,U 14590,U
BSL	REAL*8			14460,S 14550,U 14650,U
BSTL	REAL*8			14480,S 14550,U 14650,U
BSTMST	REAL*8		20	13720,S 14620,U 14630,U 14680,S
BSTRMS	REAL*8		20	13720,D 14600,U 14660,S
BSTRMT	REAL*8		20	13720,D 14610,U 14670,S
BSTTMT	REAL*8		20	13720,D 14620,U 14700,S
BSTU	REAL*8			14450,S 14550,U 14650,U
BSU	REAL*8			14430,S 14540,U 14650,U
BTL	REAL*8			14470,S 14550,U 14650,U
BXTMST	REAL*8		20	13720,D 14600,U 14690,S
BTU	REAL*8			14440,S 14540,U 14650,U
CC1	REAL*8			13660,D 14340,U 14340,U
CC2	REAL*8			13660,D 14350,U 14350,U
CHIS	REAL*8			13890,S 14160,S 14160,U 14370,U 14380,U
CHIST	REAL*8			13920,S 14240,S 14240,U 14390,U
CHIST1	REAL*8			14220,S 14240,U 14270,U
CHIS1	REAL*8			14140,S 14160,U 14250,U
CHIS2	REAL*8			14150,S 14160,U 14250,U
CHIT	REAL*8			13890,S 14190,S 14190,U 14370,U 14380,U
CHIT1	REAL*8			14170,S 14190,U 14260,U
CHIT2	REAL*8			14180,S 14190,U 14260,U
CONST	REAL*8			13610,D

STRESS

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
COSINE	REAL*8		51	13650,D 14120,U 14130,U
COSM	REAL*8		50	13650,D 14210,U 14220,U
CS	REAL*8			13980,S 14040,U 14050,U 14070,U 14090,U 14100,U 14160,U 14190,U 14270,U
CTHIS	REAL*8			13910,S 14250,S 14250,U 14500,U
CTHIST	REAL*8			13930,S 14270,S 14270,U 14510,U
CTHIT	REAL*8			13920,S 14260,S 14260,U 14500,U
CLST	REAL*8			14410,S 14430,U 14440,U 14450,U 14460,U 14470,U 14480,U
CLST	REAL*8			14420,S 14430,U 14440,U 14450,U 14460,U 14470,U 14480,U
DCDS	REAL*8			13980,U
DD1	REAL*8			13660,D 14370,U 14370,U
DD2	REAL*8			13660,D 14380,U 14380,U 14500,U 14500,U
DELTF	REAL*8			13680,D 13790,U
DSTN	REAL*8			13990,U 14610,U 14630,U
DTH	REAL*8		50,5,2	13710,D 14020,U 14020,U 14020,U
DTHY	REAL*8			14020,S 14150,U 14180,U
DT2	REAL*8			13610,D
EFS	REAL*8			13600,D
EFS	REAL*8			14300,S 14340,U 14350,U
EPST	REAL*8			14330,S 14360,U
EPT	REAL*8			14310,S 14340,U 14350,U
ES	REAL*8		5	13600,D 14040,U
EFT	REAL*8		5	13600,D 14060,U
ESTU	REAL*8			13850,S 14060,S 14060,U 14320,U
ESU	REAL*8			13830,S 14040,S 14040,U 14300,U
ESUT	REAL*8			13940,S 14090,S 14090,U 14300,U
ET	REAL*8		5	13600,D 14050,U

STRESS

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
ETIJ	REAL*8			13840,S 14050,S 14050,U 14310,U
ETU*	REAL*8			14100,S 14100,U 14310,U
E1	REAL*8		50	13640,D
E13	REAL*8		5	13600,D 14070,U 14170,U 14200,U
E13U	REAL*8			13860,S 14070,S 14070,U 14300,U 14320,U
E2	REAL*8		50	13640,D
E23	REAL*8		5	13600,D 14080,U 14170,U 14200,U
E23U	REAL*8			13870,S 14080,S 14080,U 14310,U 14320,U
ENU1	REAL*8		50	13640,D 14340,U 14370,U
ENU2	REAL*8		50	13640,D 14350,U 14380,U 14500,U
ENRCE	REAL*8		2040	13600,D
G	REAL*8		50	13640,D
G00	REAL*8			13660,D
G0M	REAL*8			13640,D
G01	REAL*8			13660,D 14360,U
G02	REAL*8			13660,D 14390,U 14510,U
HARM	REAL*8			13670,D
I	INTEGER			13820,S 13980,U 13990,U 14520,U 14600,U 14600,U 14610,U 14620,U 14620,U 14630,U 14660,U 14670,U 14680,U 14690,U 14700,U
IDCOE	INTEGER			13620,U
IDELF	INTEGER			13620,D
IH	INTEGER			13960,S 13970,U 14000,U 14010,U 14010,U 14010,U 14020,U 14020,U 14020,U 14040,U 14050,U 14060,U 14070,U 14080,U 14170,U 14170,U 14200,U 14200,U
IHARM	INTEGER		5	13670,D 13970,U
ITAM	INTEGER			13580,U 13780,U
ITAM1	INTEGER			13780,S 13810,U



STRESS

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
I1	INTEGER			13580,U 13810,U 14000,U 14010,U 14010,U 14010,U 14020,U 14020,U 14020,U 14020,U 14020,U 14020,U 14100,U 14120,U 14120,U 14130,U 14130,U 14140,U 14150,U 14170,U 14170,U 14170,U 14180,U 14200,U 14200,U 14210,U 14210,U 14210,U 14210,U 14220,U 14220,U 14220,U 14220,U 14220,U 14220,U 14340,U 14350,U 14370,U 14380,U 14410,U 14420,U 14500,U 14530,U 14540,U 14570,U 14570,U 14580,U 14580,U 14590,U 14590,U 14600,U 14600,U 14620,U 14620,U 14640,U
K	INTEGER			14000,U 14120,U 14120,U 14130,U 14130,U 14140,U 14140,U 14210,U 14210,U 14220,U
N	INTEGER			13610,D
NCLOST	INTEGER			13630,D
NLEMS	INTEGER			13610,D
NEQ	INTEGER			13610,D 14000,U
NEOT	INTEGER			13610,D
NH	INTEGER			13610,D 13960,U
NHNS	INTEGER			13610,D
NHP	INTEGER			13670,D
NM	INTEGER			13610,D
MNODES	INTEGER			13610,D
NPRNTE	INTEGER			13620,D
NPRNTL	INTEGER			13620,D
NPRNTQ	INTEGER			13610,D
NSTZF	INTEGER			13610,D
NSTRSS	INTEGER			13630,D
NTHETA	INTEGER			13630,D 13820,U
PAV	REAL*8			14580,S 14610,U 14630,U
PH	REAL*8		50	13650,D 14580,U 14580,U
PHD	REAL*8		50	13650,D 14220,U
QR3	REAL*8			14120,S 14200,U 14210,U
QR7	REAL*8			14130,S 14200,U 14210,U

STRESS

J-3.214

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
QV	REAL#R		1020	13690,D 14120,U 14120,U 14130,U 14130,U 14140,U 14140,U 14210,U 14210,U 14220,U
QV1	REAL#R		1020	13690,D
QV2	REAL#R		1020	13690,D
QP	REAL#R		1020	13690,D
QP1	REAL#R		1020	13690,D
QS	REAL#R			13690,D
R	REAL#R		50	13650,D 14170,U 14210,U 14220,U 14220,U 14570,U 14570,U 14600,U 14600,U 14620,U
RAV	REAL#R			14570,U 14600,U 14620,U
SHRS	REAL#R			14600,S 14640,U
SHRT	REAL#R			14620,S 14650,U
SINE	REAL#R		51	13640,D 14120,U 14130,U
SINM	REAL#R		50	13650,D 14170,U 14200,U 14200,U 14220,U
SN	REAL#P			13920,S 14060,U 14080,U 14240,U 14250,U 14260,U
STRESS	REAL#R			13580,D
STRMS	REAL#R			14370,S 14430,U 14460,U 14540,U 14600,U 14640,U 14660,U
STRMST	REAL#R			14390,S 14450,U 14480,U 14540,U 14620,U 14630,U 14640,U 14680,U
STRMT	REAL#R			14380,S 14440,U 14470,U 14540,U 14610,U 14640,U 14670,U
STRNS	REAL#R			14340,S 14430,U 14460,U 14540,U 14640,U
STRNST	REAL#R			14360,S 14450,U 14480,U 14540,U 14640,U
STRNT	REAL#R			14350,S 14440,U 14470,U 14540,U 14640,U
STMST	REAL#R			14510,S 14600,U 14690,U
STRMT	REAL#R			14520,S 14620,U 14700,U
T	REAL#R		50	13640,D 14410,U 14420,U
TH	REAL#P		50,5,2	13710,D 14010,U 14010,U 14010,U
THR	REAL#R			13710,D

STRESS

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
THETA	REAL*8		20	13630,D 13980,U 13990,U 14520,U
THETAS	REAL*8			13630,D
THETA1	REAL*8			14520,S 14540,U 14640,U
TMT	REAL*8			14010,S 14090,U 14100,U
TIME	REAL*8			13680,D 13790,U 14010,U 14020,U
TMFT	REAL*8			13680,D
TMI	REAL*8			13790,S 13800,U
TOTIME	REAL*8			13680,D
TPRINT	REAL*8			13800,S 13810,U
T0	REAL*8			13680,D 14010,U 14010,U 14020,U 14020,U
T1	REAL*8			13680,D 14010,U 14020,U
XIHI	REAL*8			13970,S 13980,U 13990,U 14170,U 14200,U 14200,U 14210,U 14250,U 14260,U 14270,U

TEMPCE

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
AL	REAL*8		167	16910,D 17280,U 17280,U 17290,U 17300,U 17300,U 17300,U 17310,U 17320,U 17320,U 17320,U 17330,U 17330,U 17340,U 17340,U 17350,U 17350,U 17360,U 17360,U 17360,U 17380,U 17380,U 17380,U 17390,U 17390,U 17400,U 17400,U 17420,U 17420,U
ALS	REAL*8		50	16890,U 17020,U 17050,U 17060,U 17090,U
ALT	REAL*8		50	16890,D 17030,U 17040,U 17070,U 17080,U
ARCL	REAL*8		50	16890,D 17150,U 17150,U 17160,U 17160,U 17190,U 17190,U 17210,U 17210,U
CFS	REAL*8		4	16070,D 17020,S 17040,S 17060,S 17080,S 17280,U 17280,U 17280,U 17300,U 17300,U 17310,U 17320,U 17320,U 17320,U 17330,U 17340,U 17340,U 17340,U 17350,U 17360,U 17360,U 17360,U 17380,U 17380,U 17380,U 17390,U 17400,U 17400,U 17420,U 17420,U
CHALS	REAL*8			16910,D
CHECK	REAL*8		8,8	16910,D 17510,U
CONST	REAL*8			16920,D
COSINE	REAL*8		51	16890,D
COSM	REAL*8		50	16880,D
DT4	REAL*8		50,5,2	16890,D 17270,U 17290,U 17310,U 17330,U 17350,U 17370,U 17390,U 17400,U 17420,U
DT2	REAL*8			16920,D
F1	REAL*8		50	16870,D 17020,U 17060,U
F2	REAL*8		50	16870,D 17040,U 17080,U
FNU1	REAL*8		50	16870,D 17020,U 17020,U 17060,U 17060,U 17070,U 17080,U
FNU2	REAL*8		50	16870,D 17020,U 17040,U 17040,U 17060,U 17080,U 17090,U
FORCE	REAL*8		2040	16940,D 17540,S 17540,U
G	REAL*8		50	16870,D
GFORM	REAL*8			16870,D
HARM	REAL*8			16900,D
I	INTEGER			17480,S 17490,U 17510,U 17510,U 17510,U 17520,S 17530,U 17540,U
IR	INTEGER			16850,U 17230,U 17530,U

TEORCE

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
IRP1	INTEGER			17230,S 17270,U 17270,U 17280,U 17290,U 17300,U 17310,U 17320,U 17330,U 17340,U 17350,U 17360,U 17370,U 17380,U 17390,U 17400,U 17410,U 17420,U 17430,U
IDCQF	INTEGER			16930,D
IDELF	INTEGER			16930,D
IFLM	INTEGER			16850,U 17020,U 17020,U 17020,U 17020,U 17020,U 17020,U 17030,U 17030,U 17040,U 17040,U 17040,U 17040,U 17040,U 17050,U 17050,U 17050,U 17060,U 17060,U 17060,U 17060,U 17070,U 17070,U 17070,U 17080,U 17080,U 17080,U 17080,U 17090,U 17090,U 17090,U 17100,U 17100,U 17140,U 17150,U 17150,U 17150,U 17150,U 17160,U 17160,U 17160,U 17160,U 17190,U 17190,U 17190,U 17190,U 17210,U 17210,U 17210,U 17210,U 17270,U 17270,U 17280,U 17290,U 17300,U 17310,U 17320,U 17330,U 17340,U 17350,U 17360,U 17370,U 17380,U 17390,U 17400,U 17400,U 17420,U 17420,U 17530,U
IH	INTEGER			17240,U 17250,U 17270,U 17270,U 17280,U 17290,U 17300,U 17310,U 17320,U 17330,U 17340,U 17350,U 17360,U 17370,U 17380,U 17390,U 17400,U 17410,U 17420,U 17430,U 17530,U
IHAPM	INTEGER		5	16900,D 17250,U
J	INTEGER			17460,S 17470,U 17470,U 17500,S 17510,U 17510,U 17530,U 17540,U 17540,U
KVP	INTEGER			17250,S 17260,U 17440,U
N	INTEGER			16920,D
NFLMS	INTEGER			16920,D 17140,U
NFO	INTEGER			16920,D 17530,U
NFOF	INTEGER			16920,D 17530,U
NH	INTEGER			16920,D 17240,U
NHNS	INTEGER			16920,D
NHO	INTEGER			16920,D
NH	INTEGER			16920,D
NNQDES	INTEGER			16920,D
NQRNTE	INTEGER			16920,D
NQRNTE	INTEGER			16920,D
NQRNTE	INTEGER			16920,D
NQNTQ	INTEGER			16920,D
NSIZE	INTEGER			16920,D
PH	REAL*8		50	16980,D

TFORCE

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
DHD	REAL*8		50	16880,D 17150,U 17150,U 17160,U 17160,U 17190,U 17190,U 17210,U 17210,U
DHDD	REAL*8			17170,S 17190,S 17210,S 17360,U 17390,U
DHDD1	REAL*8			17150,S 17170,U
DHDD2	REAL*8			17160,S 17170,U
DI	REAL*8			17010,S 17020,U 17040,U 17060,U 17080,U
D	REAL*8		8	16960,D 17280,S 17300,S 17320,S 17340,S 17360,S 17380,S 17400,S 17420,S 17470,S 17470,S 17510,U
DN	REAL*8		1020	16940,D
DN1	REAL*8		1020	16940,D
DN2	REAL*8		1020	16940,D
QP	REAL*8		1020	16940,D
QP1	REAL*8		1020	16940,D
QD	REAL*8		8	16960,D 17490,S 17510,S 17510,U 17540,U
QS	REAL*8			16940,D
QJES	REAL*8			16960,D
R	REAL*8		50	16880,D
SINE	REAL*8		51	16870,D
SINM	REAL*8		50	16880,D
T	REAL*8		50	16870,D 17020,U 17040,U 17060,U 17080,U
TFORCE	REAL*8			16850,D
TH	REAL*8		50,5,2	16880,D 17270,U 17280,U 17300,U 17320,U 17340,U 17360,U 17380,U 17400,U 17420,U
THR	REAL*8			16880,D
YKD	REAL*8			17260,S 17280,U 17300,U 17330,U 17350,U 17400,U 17400,U 17420,U 17420,U

THEOF

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
ALS	REAL*8		50	16150,D
ALT	REAL*8		50	16150,D
ANG	REAL*8			16330,S 16350,U
CONST	REAL*8			16120,D
DSIN	REAL*8			16610,U 16610,U 16620,U 16620,U
DTH	REAL*8		50,5,2	16150,D 16740,S
DT2	REAL*8			16120,D
FRCE	REAL*8			16140,D
HARM	REAL*8			16160,D
I	INTEGER			16300,U 16300,U 16300,U 16300,S 16390,U 16390,U 16390,U 16400,U 16400,S 16520,S 16530,U 16530,U 16530,U 16540,U 16540,U 16540,U 16580,S 16590,U 16600,U 16610,U 16620,U
IR	INTEGER			16100,U 16720,U
IRP1	INTEGER			16720,S 16730,U 16740,U
JDOCF	INTEGER			16130,D
IOELE	INTEGER			16130,D
IDP	INTEGER			16420,S 16430,U 16430,U
IFLM	INTEGER			16100,U 16220,U 16240,U 16240,U 16250,U 16730,U 16740,U
IFLM1	INTEGER			16320,S 16390,U
IFLM2	INTEGER			16220,S 16240,U 16300,S 16390,U
IH	INTEGER			16450,S 16460,U 16730,U 16740,U
IHARM	INTEGER		5	16160,D 16460,U
KEY	INTEGER			16340,S 16350,U 16360,U 16370,U
KYP	INTEGER			16460,S 16470,U 16510,U 16660,U
LE	INTEGER			16240,U
N	INTEGER			16120,D
NO	INTEGER			16170,D 16300,U

THCDE

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
N00	INTEGER			16300,S 16300,U 16310,U 16320,U 16330,U 16400,U 16410,U 16500,U 16520,U 16580,U
N002	INTEGER			16310,S 16340,U
N02	INTEGER			16410,S 16420,U
N0ELEM	INTEGER			16120,D
N00	INTEGER			16120,D
N0QT	INTEGER			16120,D
NH	INTEGER			16120,D 16450,U
NHNS	INTEGER			16120,D
NHP	INTEGER			16160,D
NN	INTEGER			16120,D
NN00ES	INTEGER			16120,D
NPRNTE	INTEGER			16130,D
NPRNTE	INTEGER			16130,D 16250,U 16290,U
NPRNTEQ	INTEGER			16120,D
NS	INTEGER			16170,D
NSIZE	INTEGER			16120,D
NT	INTEGER			16170,D
P	REAL*8		74	16140,D 16300,S 16360,S 16360,U 16390,U 16530,U 16610,U 16670,U
PI	REAL*8			16230,S 16550,U 16560,U 16630,U 16640,U
PINT	REAL*8			16480,S 16530,S 16530,U 16550,S 16550,U 16610,S 16610,U 16630,S 16630,U 16670,S 16700,U 16730,U
R	REAL*8		74	16140,D 16300,S 16370,S 16370,U 16390,U 16540,U 16620,U 16680,U
RINT	REAL*8			16490,S 16540,S 16540,U 16560,S 16560,U 16620,S 16620,U 16640,S 16640,U 16680,S 16710,S 16740,U
S	REAL*8		74	16140,D
TAPES	REAL*8			16170,D
TH	REAL*8		50,5,2	16150,D 16730,S



THCOF

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
THCOF	REAL*8			16100,D
THCR	REAL*8			16150,D
THCTR	REAL*8		74	16140,D 16300,S 16330,U 16350,S 16390,U 16390,U 16430,S 16430,U 16530,U 16530,U 16540,U 16540,U 16590,U 16600,U
X1	REAL*8			16590,S 16610,U 16620,U
X2	REAL*8			16600,S 16610,S 16620,S
YK0	REAL*8			16470,S 16590,U 16600,U 16610,U 16620,U

TRJ4OR

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
CCC	REAL*8		125	17600,D 17950,S
CCCC	REAL*8		625	17610,D 18440,S
CONST	REAL*8			17620,D
CS	REAL*8			17600,D
CSS	REAL*8		125	17600,D 17970,S
CS4	REAL*8			17610,D
DT2	REAL*8			17620,D
FFIGHT	REAL*8			18230,S 18390,S 18430,S 18470,U
FFIVE	REAL*8			18200,S 18360,S 18400,S 18470,U
FFOUR	REAL*8			17860,S 17930,S 17940,S 17970,U 18190,S 18300,S 18310,S 18440,U 18450,U 18460,U
CONF	REAL*8			17830,S 17870,S 17880,S 17950,U 17960,U 18160,S 18240,S 18250,S 18440,U 18450,U
FSEVEN	REAL*8			18220,S 18380,S 18420,S 18470,U
FSIX	REAL*8			18210,S 18370,S 18410,S 18470,U
ETHREE	REAL*8			17850,S 17910,S 17920,S 17970,U 18180,S 18280,S 18290,S 18440,U 18450,U 18460,U
ETWO	REAL*8			17840,S 17890,S 17900,S 17950,U 17960,U 18170,S 18260,S 18270,S 18440,U 18450,U
HARM	REAL*8			17640,D
IARS	INTEGER			17780,U 17820,U 18080,U 18120,U 18330,U 18350,U
IDCOF	INTEGER			17630,D
IDFLF	INTEGER			17630,D
IFD	INTEGER			18000,S 18150,S 18150,U 18440,U 18450,U 18460,U 18470,U
IHARM	INTEGER		5	17640,D 17720,U 17740,U 17760,U 18020,U 18040,U 18060,U 18100,U
II	INTEGER			17740,S 17770,U 17780,U 17810,U 17820,U 18040,S 18070,U 18080,U 18340,U 18350,U
IMJ	INTEGER			17780,S 17790,U 17800,U 17900,U 18080,S 18130,U 18130,U 18280,U 18290,U 18300,U

TRT4OR

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
IML	INTEGER			18350,S 18360,U 18370,U 18400,U 18410,U
IMM	INTEGER			17820,S 17930,U 17940,U
IPJ	INTEGER			17770,S 17790,U 17870,U 17880,U 18070,S 18130,U 18130,U 18240,U 18250,U 18260,U 18270,U
IPL	INTEGER			18340,S 18380,U 18390,U 18420,U 18430,U
IPM	INTEGER			17810,S 17910,U 17920,U
ITM	INTEGER			17710,S 17800,S 17800,U 17950,U 17960,U 17970,U
IJ	INTEGER			17760,S 17770,U 17780,U 17910,U 17920,U 17920,U 17930,U 17940,U 17940,U 18060,S 18070,U 18080,U 18320,U 18330,U
KK	INTEGER			18100,S 18110,U 18120,U 18320,U 18330,U
KMJ	INTEGER			18330,S 18370,U 18390,U 18410,U 18410,U 18430,U 18430,U
KML	INTEGER			18120,S 18130,U 18130,U 18260,U 18270,U 18270,U 18300,U 18310,U 18310,U
KPJ	INTEGER			18320,S 18360,U 18380,U 18600,U 18400,U 18420,U 18420,U
KPL	INTEGER			18110,S 18130,U 18130,U 18240,U 18250,U 18250,U 18280,U 18290,U 18290,U
LI	INTEGER			18020,S 18110,U 18120,U 18340,U 18350,U
NLEMS	INTEGER			17620,D
NEQ	INTEGER			17620,D
NEQT	INTEGER			17620,D
NH	INTEGER			17620,D 17710,U 17730,U 17750,U 18010,U 18030,U 18050,U 18090,U
NHNS	INTEGER			17620,D
NHP	INTEGER			17640,D
NM	INTEGER			17620,D
NNDOFS	INTEGER			17620,D
NPRNTE	INTEGER			17630,D
NPRNTL	INTEGER			17630,D
NPRNTQ	INTEGER			17620,D
NSIZE	INTEGER			17620,D

TRI40R

VARIABLE	TYPE	INITIAL VALUE	DIMENSION	WHERE/HOW USED
PT02	REAL*8			17690,S 17950,U 17960,U 17970,U
PT04	REAL*8			17990,S 18440,U 18450,U 18460,U 18470,U
SC05	REAL*8		625	17610,D 18470,U
SS0	REAL*8		125	17600,D 17960,S
SS00	REAL*8		625	17610,D 18460,S
SSSS	REAL*8		625	17610,D 18450,S
TRI40R	REAL*8			17580,D

COMMON MAP

COMMON BLOCKS	ROUTINES														
	MAIN	INPUT	SETUP	NLTERM	QPRIME	HOUBQ1	HOUBQN	NRESTR	MATMUT	SOLVEQ	STRESS	FRCES	THCOE	TFORCE	TRI40R
CHALS		X													
CONST	X	X	X	X	X	X	X	X		X	X	X	X	X	X
CS					X										X
CS4					X										X
CYCLE	X	X								X					
EES				X	X						X				
FRCE												X	X		
GCD			X	X	X						X				
GEOM		X		X	X						X			X	
HARM	X	X	X		X						X	X	X	X	X
NLTRMS				X	X										
PRINT	X	X	X	X						X					
PS		X				X	X								
QS	X	X	X	X	X	X	X				X				X
QUES												X			X
RESTRT	X	X	X			X									
RSTRNT		X				X	X	X							
RZ		X			X										
SLVEEQ		X	X			X	X	X		X					
TAPES	X	X										X	X		
THCON		X			X										
THER		X			X						X		X		
THETAS		X	X	X							X			X	
TMFT	X	X		X	X	X	X				X				

## EQUIVALENCES

FRCES

EQUIVALENT VARIABLES

NONE

HOUBQN

EQUIVALENT VARIABLES

NONE

HOUBQ1

EQUIVALENT VARIABLES

NONE

INPUT

EQUIVALENT VARIABLES

DUM(1),XN(1)  
XN(1),COMENT(1)

MAIN

EQUIVALENT VARIABLES

QN(1),CARD(1)

MATMUT

EQUIVALENT VARIABLES

NONE

NLTERM

EQUIVALENT VARIABLES

NONE

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EQUIVALENCES (continued)

NRESTR

EQUIVALENT VARIABLES

NONE

QPRIME

EQUIVALENT VARIABLES

NONE

SETUP

EQUIVALENT VARIABLES

NONE

SOLVEQ

EQUIVALENT VARIABLES

NONE

STRESS

EQUIVALENT VARIABLES

NONE

TFORCE

EQUIVALENT VARIABLES

NONE

THCOE

EQUIVALENT VARIABLES

NONE

EQUIVALENCES (continued)

TRI40R

EQUIVALENT VARIABLES

NONE



FRCES

## FORMAT STATEMENT MAP

LOCATION	STATEMENT NUMBER	SPECIFICATION
15980	160	1H1,35X,49HFOURIER COEFFICIENTS OF APPLIED PRESSURE LOADINGS,/,20X,10HMERIDIONAL,20X,6HNORMAL,20X,10HTANGENTIAL,10X,12HHARMONIC NO.,/WRITE (6,160)
16010	170	2I5/(3F10.0) READ(ND,170)IELM1,IELM2,(P(I),R(I),S(I),I=1,NH)
16020	180	/60X,11HELEMENT NO.,I3,1H-,I3, /(2X,3D28.7,15X,I2) WRITE(6,180)IELM1,IELM2,(P(I),R(I),S(I),I=1,NH)
	190	1H1,51X,30HAPPLIED LOADS ON THE STRUCTURE///56X,19HPRESSURE COMPONENTS //20X,10HMERIDIONAL,20X,6HNORMAL,20X,10HTANGENTIAL,11X,19HFROM THETA TO THETA,9H(DEGREES) WRITE(6,190)
16060	200	3I5/(4F10.0) READ(ND,200)IELM1,IELM2,NDP,(THETB(I),P(I),R(I),S(I),I=1,NDP)
16070	210	/60X,11HELEMENT NO.,I3,1H-,I2/(2X,3F28.3,12X,2F10.3) WRITE(6,210)IELM1,IELM2,(P(I),R(I),S(I),THETB(I),THETB(I+1),I=1,NDP)
16070	220	2X,3F28.3,12X,2F10.3 WRITE(6,220)(P(I),R(I),S(I),THETB(I-1),THETB(I),I=NDPP2,ND2)

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## FORMAT STATEMENT MAP

LOCATION	STATEMENT NUMBER	SPECIFICATION
05050	730	1H1,38X,65HDYNASOR-II - DYNAMIC NONLINEAR ANALYSIS OF SHELLS OF REVOLUTION// WRITE(6,730)
05070	740	2I5 READ(ND,740)NCARDS,NT
05080	750	20A4 READ(ND,750)(COMENT(J),J=1,20)
05090	760	2F10.0,4I5,/,10I5 READ(ND,760)TOTIME,DELTE,IRSTRT,INCRST, NCLOSE,ITELF,NPRNTQ,IPRINT,NCLCST, NSTRSS,NPRNT,NPRNIT,NPRNTL,NPRNTF, NPRNTH,NPRNMS
05100	770	I5,/, (8F10.0) READ(ND,770)NTHETA,(THETA(I),I=1,NTHETA)
05110	780	16I5 READ(ND,780)NODRES READ(ND,780)NP,NDIRCT READ(ND,780)IQN,IQNT
05120	790	///,2X,46H**SHELL IDENTIFICATION COMMENTS FROM SAMMSOR** WRITE(6,790)
05130	800	/5X,20A4 WRITE(6,800)(COMENT(J),J=1,20)
05140	810	1H1,50X,33HCONTROL CONSTANTS AND COMMENTS///35X,8HTOTIME =,F12.9,22X, 7HDELTE =,F13.9/35X,8HIRSTRT =,I12, 22X,8HINCRST =,I12/35X,7HNPRNT =, I13,22X,8HNPRNIT =,I12/35X,8HNPRNTQ =, I12,22X,8HNPRNIT =,I12/35X,8HNCLCST =, I12,22X,8HNSTRSS =,I12/35X,8HNPRNTL =, I2,22X,8HNPRNTF =,I12/35X,8HNPRNTH =, I12,22X,4HNT =,I16/35X,4HNS =,I16,22X, 4HND =,I16,/35X,8HNCLOSE =,I12,22X, 7HITELF =,I13/35X,8HNELEMS =,I12,22X, 8HNPRNMS =,I12/35X,4HNNH =,I16,/35X, 7HIHARM =,5I11//

## INPUT

## FORMAT STATEMENT MAP

LOCATION	STATEMENT NUMBER	SPECIFICATION
		WRITE(6,810)TOTIME,DELTE,IRSTRT, INCRST,NPRNT,NPRNIT,NPRNTQ,IPRIT, NCLCST,NSTRSS,NPRNTL,NPRNTF,NPRNTF, NPRNTH,NT,NS,ND,NCLOSE,ITELF,NELEMS, NPRNMS,NH,(IHARM(I),I=1,NH)
05210	820	35X,8HNTHETA =,I12,/35X,7HTHETA =, 5F10.2,(/,42X,5F10.2) WRITE(6,820)NTHETA,(THETA(I),I=1,NTHETA)
05220	830	/////50X,29HNUMBER OF NODAL RESTRAINTS ISI5//52X,9HDIRECTION,12X,7HAPPLIES, //,57X,1H1,10X,15HAXIAL RESTRAINT,/, 57X,1H2,10X,20HTANGENTIAL RESTRAINT, /,57X,1H3,10X,16HRADIAL RESTRAINT,/, 57X,1H4,10X,17HANGULAR RESTRAINT,//, 58X,15HNODE DIRECTION/ WRITE(6,830)NODRES
05260	840	58X,I3,7X,I1 WRITE(6,840)NP,NDIRCT
05270	850	2I5,4F10.0 READ(ND,850)IN1,IN2,Q1,Q2,Q3,Q4 READ(ND,850)IN1,IN2,Q1,Q2,Q3,Q4
05280	860	1H1,7X,7HINITIAL,29X,10HVELOCITIES,22X, 3HAND,19X,13HDISPLACEMENTS//4X, 124HNODE HARMONIC AXIAL TANGENTIAL RADIAL ANGULAR AXIAL TANGENTIAL RADIAL ANGULAR // WRITE(6,860)
05320	870	5X,I2,6X,I2,3X,8D14.4 WRITE(6,870)II,IHARM(I),QN(IQ+IX+1), QN(IQ+IX+2),QN(IQ+IX+3),QN(Q+IX+4), QN1(IQ+IX+1),QN1(IQ+IX+2),QN1(IQ+IX+3), QN1(IQ+IX+4)
05330	880	2I5,2F10.0 READ(ND,880)IELM1,IELM2,ALS11,ALTI1

## INPUT

## FORMAT STATEMENT MAP

LOCATION	STATEMENT NUMBER	SPECIFICATION
05340	890	1H1,45X,41HELEMENT ELASTIC AND GEOMETRIC PROPERTIES,///67HELEMENT ALPHA--S ALPHA--T E1 E2 FNU1 FNU2 G,11X,1HR,11X,1HT,9X,4HARCL,9X,2HPH, 10X,3HPHP// WRITE(6,890)
05370	900	3X,12,2X,4D10.2,2F6.3,6D12.4 WRITE(6,900)(I,ALS(I),ALT(I),E1(I),E2(I), FNU1(I),FNU2(I),B(I),R(I),T(I),ARCL(I), PH(I),PHP(I),I=1,NELEMS)
05380	910	1H1,38X,15HHARMONIC NUMBER,I5,37H HAS THE FOLLOWING STIFFNESS MATRIX// WRITE(6,910)IHARM(JH)
05400	920	2X,D16.8,/,2X,2D16.8,/,2X,3D16.8,/,2X, 4D16.8,/,2X,5D16.8,/,2X,6D16.8,/, 2X,7D16.8,/,2X,8D16.8,/, (2X,5D16.8,/, 2X,6D16.8,/,2X,7D16.8,/,2X,8D16.8,/) ) WRITE(6,920)(XN(I+NN),I=1,NSIZE) WRITE(6,920)(XP(I+NN),I=1,NSIZE)
05430	930	1H1,38X,15HHARMONIC NUMBER,I5,32H HAS THE FOLLOWING MASS MATRIX// WRITE(6,930)IHARM(JH)
05450	940	1H1/////5X,45HTHIS SOLUTION STARTS AFTER TIME INCREMENT NO.,I5,19H WHERE THE TIME WAS,F12.4,13H MICROSECONDS,/,5X, 27H AND THE TIME INCREMENT WAS,D12.5/////
05480	950	F10.0,4I5,A8 READ(ND,950)T1,NCF,IDELF,IDCOE,ITCOE,CONSTF
05490	960	40H1FOLLOWING IS LOAD DESCRIPTION AT TIME =, F12.4,13H MICROSECONDS,5X,A8 WRITE(6,960)TPRNT,CONSTF
05510	970	2I5,4F10.0 READ(ND,970)NCF1 READ(ND,970)IN1,IN2,F1,F2,F3,F4

## INPUT

## FORMAT STATEMENT MAP

LOCATION	STATEMENT NUMBER	SPECIFICATION
05520	980	///20X,30HCONCENTRATED FORCES HARMONIC , I5//6X,8HNODE NO.,6X,5HAXIAL,10X, 10HTANGENTIAL,10X,6HRADIAL,13X, 7HANGULAR/ WRITE(6,980)IHARM(IH)
05540	990	I10,4D20.8 WRITE(6,990)IN,F1,F2,F3,F4
05550	1000	2I5,/, (2F10,0) READ(ND,1000)IELM1,IELM2,(TH1(IH),DTH1(IH), IH=1,NH)
05560	1010	1H1,25X,39HTEMPERATURE COEFFICIENTS, HARMONIC NO. I3//10X,11HELEMENT NO., 17X,12HTEMP. COEFF.,12X,18HTEMP. GRAD. COEFF./// WRITE(6,1010)IHARM(IH)
05580	1020	I20,2D30.5 WRITE(6,1020)IELM,TH(IELMIN,IBP1), DTH(IELM,IH,IBP1)
05590	1030	1H1,25X,32HGENERALIZED FORCES, HARMONIC NO.,I3,//6X,8HNODE NO.,6X,5HAXIAL, 13X,10HTANGENTIAL,11X,6HRADIAL,13X, 7HANGULAR/// WRITE(6,1030)KYP
05610	1040	I9,4D19.8 WRITE(6,1040)I,FORCE(K+1),FORCE(K+2), FORCE(K+3),FORCE(K+4)
05620	1050	1H1/////5X,42HRESTART INFORMATION FOR TIME INCREMENT NO.,I5,/,10X,22H CORRESPONDING TO TIME,F12.4,13H MICROSECONDS,/,2X,51H HAS BEEN PLACED ON TAPE FOR USE IN SUBSEQUENT RUNS// WRITE(6,1050)ITAM,TPRNT

## MAIN

## FORMAT STATEMENT MAP

LOCATION	STATEMENT NUMBER	SPECIFICATION
00790 00200	110	3I5 READ (5,110) NCASES,ND,NS
00800 00210	120	1H1,///,30X,31H THE NUMBER OF CASES TO BE RUN=I5 WRITE (6,120) NCASES
00810 00300	130	20A4 READ (5,130) CARD
00820 00350	140	//8H1 NCASE=,I1//,28X,22H PRINTOUT OF INPUT DATA, WRITE (6,140) NCASE
00830 00360	150	13X,2H10,8X,2H20,8X2H30,8X2H40,8X,2H50, 8X,2H60,8X,2H70,8X,2H80/5X,80H1234567 8901234567890123456789012345678901234 567890123456789012345678901234567890, WRITE (6,150)
00860 00390	160	5X,20A4 WRITE (6,160) CARD
00870 00430	170	72H THE NUMBER OF INPUT CASES DOES NOT AGREE WITH THE VALUE OF NCASES INPUT WRITE (6,170)
00890 00760	180	1H1//10X,18H ALL DATA PROCESSED//10X, 11H...STOP WRITE (6,180)

## SETUP

## FORMAT STATEMENT MAP

LOCATION	STATEMENT NUMBER	SPECIFICATION
06430	110	1H1,30X,6HITAM =,I5,5X,6HTIME =,F12.4, 13H MICROSECONDS// WRITE(6,110)ITAM,TPRNT WRITE(6,110)ITAM,TPRNT
06440	120	36X,22HDISPLACEMENTS OF NODES/38X, 9HHARMONIC ,I5//6X,8HNODE NO.,6X, 5HAXIAL,13X,10HTANGENTIAL,11X, 6HRADIAL,13X,7HANGULAR// WRITE(6,120)KY
06460	130	I10,4D20.8 WRITE(6,130)I,(QN(K+J),J=1,4) WRITE(6,130)I,(QLOAD(K+J),J=1,4)
06470	140	25X,34HDISPLACEMENTS OF NODES AT THETA =, F8.3,9H DEGREES/38X,13HALL HARMONICS/2X, 8HNODE NO.,9X,5HAXIAL,12X,10HTANGENTIAL, 12X,6HRADIAL,13X,7HANGULAR// WRITE(6,140)THETA1
06500	150	1H1,5X,4HITAM,I5,5X,4HTIME,E12.5//6X, 55HEXECUTION TERMINATED - DISPLACEMENTS GREATER THAN 1.E+4 WRITE(6,150)ITAM,TIME

STRESS

FORMAT STATEMENT MAP

LOCATION	STATEMENT NUMBER	SPECIFICATION
14740	50	<p>1H1,3X,6HITAM =,I5,3X,6HTIME =,F12.4,            13H MICROSECONDS,//,47X,33HSTRESSES            AND STRESS RESULTANTS,/,25X,17HFORCE            RESULTANTS,31X,18HMOMENT RESULTANTS,            18X,17HSHEAR RESULTANTS,/,19X,            109HN(S)            N(T) N(ST) M(S) M(T)            M(ST) Q(S) Q(T),/,12H ELEM THETA,            /,104H            NO (DEG) ***** OUTER SURFACE STRESSES *****            INNER SURFACE STRESSES *****/,15X,88H* SIGMA(S) S            GMA(T) SIGMA(ST) SIGMA(S) SIGMA(T) SIGMA(ST            ) */)            WRITE(6,50)ITAM1,TPRINT</p>
14830	60	<p>I4,F8.2,6(1PD15.4),30H XXXX XXXX ,/,12H,            STRESSES **,6(1PD15.4)            WRITE(6,60)I1,THETA1,STRNS,STRNT,STRNST,            STRMT,STRMST,BSU,BTU,BSTU,BSL,BTL,BSTL</p>
14850	70	<p>I4,F8.2,8(1PD15.4),/,12H STRESSES **,            6(1PD15.4)            WRITE(6,70)I1,THETA1,STRNS,STRNT,STRNST,            STRMS,STRMT,STRMST,SHRS,SHRT,BSU,BTU,            BSTU,BSL,BTL,BSTL</p>



THCOE

## FORMAT STATEMENT MAP

LOCATION	STATEMENT NUMBER	SPECIFICATION
16780	110	1H1,41X,47HTEMPORATURES AND THERMAL GRADIENTS ON STRUCTURE//27X, 11HTEMPORATURE,10X,16HTHERMAL GRADIENT,10X,29HFROM THETA TO THETA (DEGREES)//
16810	120	3I5/(3F10.0) READ(ND,120)IELM1,IELM2,NDP,(THETB(I), P(I),R(I),I=1,NDP)
16820	130	/,60X,11HELEMENT NO.,I3,1H-,I2,/,/ (28X,F9.3,15X,F10.3,16X,1F7.2,2X, F7.2) WRITE(6,130)IELM1,IELM2,(P(I),R(I), THETB(I),THETB(I+1),I=1,NDP)

## LABEL CROSS REFERENCE MAP

There is a label or statement number cross reference map listed for each routine. This listing gives an ascending statement number listing with corresponding references to that number. The listing gives the statement number being referenced, sequence number of referencing statements, and a corresponding letter value for each statement reference. The letter values for each reference are one of the following:

- L - this letter indicates that the referencing statement is a DO LOOP and the statement number given is the lower bound for the loop.
- B - this indicates that the referencing statement is a branch to the given statement number.
- R - an R indicates that the statement number listed is the label for a READ statement.
- W - this indicates that the statement number listed is the label for a WRITE statement.

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## FRCS

## LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCES
10	15050,B
20	15190,L
30	15330,L
40	15040,B 15110,B
50	15400,L
60	15470,B
70	15440,B
80	15560,B
90	15510,B 15550,B 15600,B
100	15430,L
110	15390,B
120	15720,B
130	15880,L 15900,L
140	15920,L
150	15360,L
160	15050,L
170	15080,R
180	15090,W
190	15130,W
200	15170,R
210	15210,W
220	15310,W

HOUBQN

## LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCES
10	9920,L
20	9990,L
30	10120,L
40	10110,B
50	10190,L

HOUBQ1

LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCES
10	9270,B
20	9180,L 9230,B
30	9320,B
40	9300,L 9340,B
50	9380,L
60	9600,B
70	9470,L
80	9520,L
90	9460,B
100	9630,L
110	9670,L

## INPUT

## LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCES
10	1180,B
20	1250,L
30	1240,B
40	1760,B
50	1480,L 1500,B
60	1460,B
70	1580,L
80	1630,B
90	1620,L 1650,B
100	1720,L
110	1540,B
120	1930,L
130	2020,L
140	2140,L
150	2130,B
160	2370,B
170	2310,L
180	2270,L
190	2260,B
200	2510,B
210	2450,L
220	2410,L
230	2400,B
240	2550,L 2570,L
250	2060,B
260	2640,L
270	2710,B
280	2700,L 2730,B
290	2780,L
300	2910,B
310	2880,L
320	2820,B
330	2930,L
340	2920,B
350	3050,L
360	3060,B
370	3040,L 3100,B 3150,B
380	3350,B
390	3370,L
400	3220,B



## INPUT

## LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCE
810	1780,W
820	1810,W
830	2120,W
840	2170,W
850	2300,R 2440,R
860	2540,W
870	2590,W
880	2870,R
890	3010,W
900	3020,W
910	3160,W
920	3170,W 3190,W
930	3180,W
940	3510,W
950	3980,R
960	4060,W
970	4110,R 4160,R
980	4140,W
990	4230,W
1000	4420,R
1010	4560,W
1020	4580,L
1030	4670,W
1040	4700,W
1050	5020,W



MAIN

LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCES
10	410,B
20	400,B
30	340,B
40	310,B
50	750,B
60	510,L
70	600,B
80	640,B 650,B
90	620,L
100	700,B
110	200,R
120	210,W
130	300,R 330,W
140	350,W
150	360,W
160	390,A
170	400,W
180	750,A

MATMUT

## LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCES
10	10940,L
20	11010,L 11010,L
30	11160,L
40	11110,L 11140,L

NLTERM

LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCE
10	6730,L
20	6910,B
30	6950,L 6970,L
40	6760,L

NRESTR

## LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCES
10	10430,L
20	10420,L
30	10410,B
40	10560,L
50	10610,L
60	10690,L
70	10520,B 10660,B
80	10380,L
90	10370,B
100	10800,L

QPRIME

LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCE
10	7590,L
20	7980,L
30	8010,L 8030,L 8070,B
40	7910,L
50	8370,L 8390,L 8430,L 8470,B
60	8330,L
70	8720,L 8740,L 8780,B
80	8680,L
90	8650,B

## SETUP

## LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCE
10	5900,B
20	5960,L
30	6060,L
40	6170,B
50	6080,L 6130,L 6160,L 6190,B
60	6250,L
70	6050,L
80	6000,B 6010,B
90	5830,L 5910,B 5920,B 6330,B
100	6380,B
110	5940,W 6220,W
120	5950,W
130	5980,W 6270,W
140	6240,W
150	6350,W

SOLVEQ

LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCES
10	11390,L
20	11420,L
30	11640,B
40	11640,B
50	11710,B
60	12590,B
70	12590,B
80	12670,B
90	
100	11440,B
110	13090,L
120	13040,B
130	13310,B
140	13310,B
150	13280,L 13400,B
160	13500,L
170	13530,L

## STRESS

## LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCES
10	13960,L
20	14530,B
30	14560,B
40	13820,L
50	13810,W
60	14540,W
70	14640,W



## TFORCE

## LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCES
10	17130,B
20	17140,B
30	17180,B
40	17200,B 17440,B
50	17460,L
60	17450,B
70	17480,L 17500,L
80	17520,L
90	17240,L 17270,B

THCOE

## LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCES
10	16320,L
20	16420,L
30	16240,B
40	16520,L
50	16510,B
60	16580,L
70	16500,B
80	16660,B
90	16570,B 16650,B 16690,B
100	16450,L
110	16250,W
120	16300,R
130	16390,W

TRI40R

LABEL CROSS REFERENCE MAP

STATEMENT NUMBER	REFERENCES
10	17710,L 17730,L 17750,L 17790,B
20	18010,L 18030,L 18050,L 18090,L 18130,B

## FRCES

## SUBROUTINES CALLED

NAME	LOCATION
NONE	

## CALLED BY ROUTINES

INPUT

## HOUBQN

## SUBROUTINES CALLED

NAME	LOCATION
MATMUT(IH,QN2,XP,QLOAD,NEQ)	09960
SOLVEQ(IH)	10180

## CALLED BY ROUTINES

SETUP

## HOUBQ1

## SUBROUTINES CALLED

NAME	LOCATION
MATMUT(IH,QN,XP,QLOAD,NEQ)	09170
MATMUT(IH,QN,XP,QLOAD,NEQ)	09280
NRESTR(KY)	09430
SOLVEQ(IH)	09450
NRESTR(KY)	09740

## CALLED BY ROUTINES

SETUP

INPUT

SUBROUTINES CALLED

NAME	LOCATION
TRI4OR	03210
NLTERM(0)	03360
FRCES(IELM,ALPHK,IB)	04320
THCOE(IELM,IB)	04510
TFORCE(IELM,IB)	04520

CALLED BY ROUTINES

MAIN  
SETUP

MAIN

SUBROUTINES CALLED

NAME	LOCATION
INPUT(1)	0470
INPUT(2)	0670
NLTERM(ITAM)	0740
SETUP(ITAM,TIME,LARGE)	0690

CALLED BY ROUTINES

NONE

MATMUT

SUBROUTINES CALLED

NAME	LOCATION
NONE	

CALLED BY ROUTINES

HOUBQ1  
HOUBQN

## NLTERM

## SUBROUTINES CALLED

NAME	LOCATION
QPRIME(I1)	06870
STRESS(I1,ITAM)	06930

## CALLED BY ROUTINES

MAIN  
INPUT  
SETUP

## NRESTR

## SUBROUTINES CALLED

NAME	LOCATION
NONE	

## CALLED BY ROUTINES

HOUBQ1

## QPRIME

## SUBROUTINES CALLED

NAME	LOCATION
NONE	

## CALLED BY ROUTINES

NLTERM

SETUP

SUBROUTINES CALLED

NAME	LOCATION
NLTERM(ITAM)	05820
HOUBQN(KY,IH)	05870
HOUBQ1(KY,IH)	05880
INPUT(3)	06400

CALLED BY ROUTINES

MAIN

SOLVEQ

SUBROUTINES CALLED

NAME	LOCATION
NONE	

CALLED BY ROUTINES

HOUBQ1  
HOUBQN

STRESS

SUBROUTINES CALLED

NAME	LOCATION
NONE	

CALLED BY ROUTINES

NLTERM

## TFORCE

## SUBROUTINES CALLED

NAME

LOCATION

NONE

## CALLED BY ROUTINES

INPUT

## THCOE

## SUBROUTINES CALLED

NAME

LOCATION

NONE

## CALLED BY ROUTINES

INPUT

## TRI40R

## SUBROUTINES CALLED

NAME

LOCATION

NONE

## CALLED BY ROUTINES

INPUT



FRCES

## SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
NONE		

HOUBQN

## SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
IH	MATMUT	09960
IH	SOLVEQ	10180
NEQ	MATMUT	09960
QLOAD	MATMUT	09960
QN2	MATMUT	09960
XP	MATMUT	09960

HOUBQ1

## SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
IH	MATMUT	09170
IH	MATMUT	09280
IH	SOLVEQ	09450
KY	NRESTR	09430
KY	NRESTR	09740
NEQ	MATMUT	09170
NEQ	MATMUT	09280
QLOAD	MATMUT	09170
QLOAD	MATMUT	09280
QN	MATMUT	09170
QN	MATMUT	09280
XP	MATMUT	09170
XP	MATMUT	09280

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## INPUT

## SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
ALPHK	FRCES	04320
IB	FRCES	04320
IB	THCOE	04510
IB	TFORCE	04520
IELM	FRCES	04320
IELM	THCOE	04510
IELM	TFORCE	04520
O	NLTERM	03360

## MAIN

## SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
ITAM	SETUP	00690
ITAM	NLTERM	00740
LARGE	SETUP	00690
TIME	SETUP	00690

## MATMUT

## SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
NONE		

## NLTERM

## SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
ITAM	STRESS	06930
I1	QPRIME	06870
I1	STRESS	06930

## NRESTR

## SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
NONE		

## QPRIME

## SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
NONE		

## SETUP

## SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
IH	HOUBQN	05870
IH	HOUBQ1	05880
ITAM	NLTERM	05820
KY	HOUBQN	05870
KY	HOUBQ1	05880
3	INPUT	06400

## SOLVEQ

## SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
NONE		

STRESS

SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
NONE		

TFORCE

SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
NONE		

THCOE

SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
NONE		

TRI4OR

SUBROUTINE CALLING ARGUMENTS

VARIABLE	SUBROUTINE	LOCATION
NONE		

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## SECTION III

## PROGRAM INPUT

The DYNASOR II code has been written so that the code can be employed by researchers who are not familiar with the inner workings of the program. Utilizing the guidelines and adhering to the limitations presented in the previous section, it is believed that most users will find it relatively easy to employ the code.

The code is available in the FORTRAN IV language using double precision or single precision arithmetic. This double precision version requires a storage space of about 330K bytes on IBM 360/65 system while the single precision storage space is about 200K bytes. Efforts have been made to make this code compatible with a large number of computing systems. In particular, adaption of the code for use on a CDC 6600 computer requires only minor changes.

The input data for a run consists of one card I (card types will be explained on the following pages) followed by a complete set of data (cards II-X) for each case. The set of cards II-X is the input data required to generate the response of a shell for a given number of harmonics due to a particular loading. The cards comprising the data deck for both an initial run and a restart are schematically represented in Fig. 1. The cards specifying the Fourier harmonics, the initial conditions, and the boundary conditions are omitted from the input deck when using the restart mode. If more than one case is to be run, include a set of data for each of the cases. There is no limit on the number of cases which may be included in a run. A card must be placed at the end of the data for the final case.

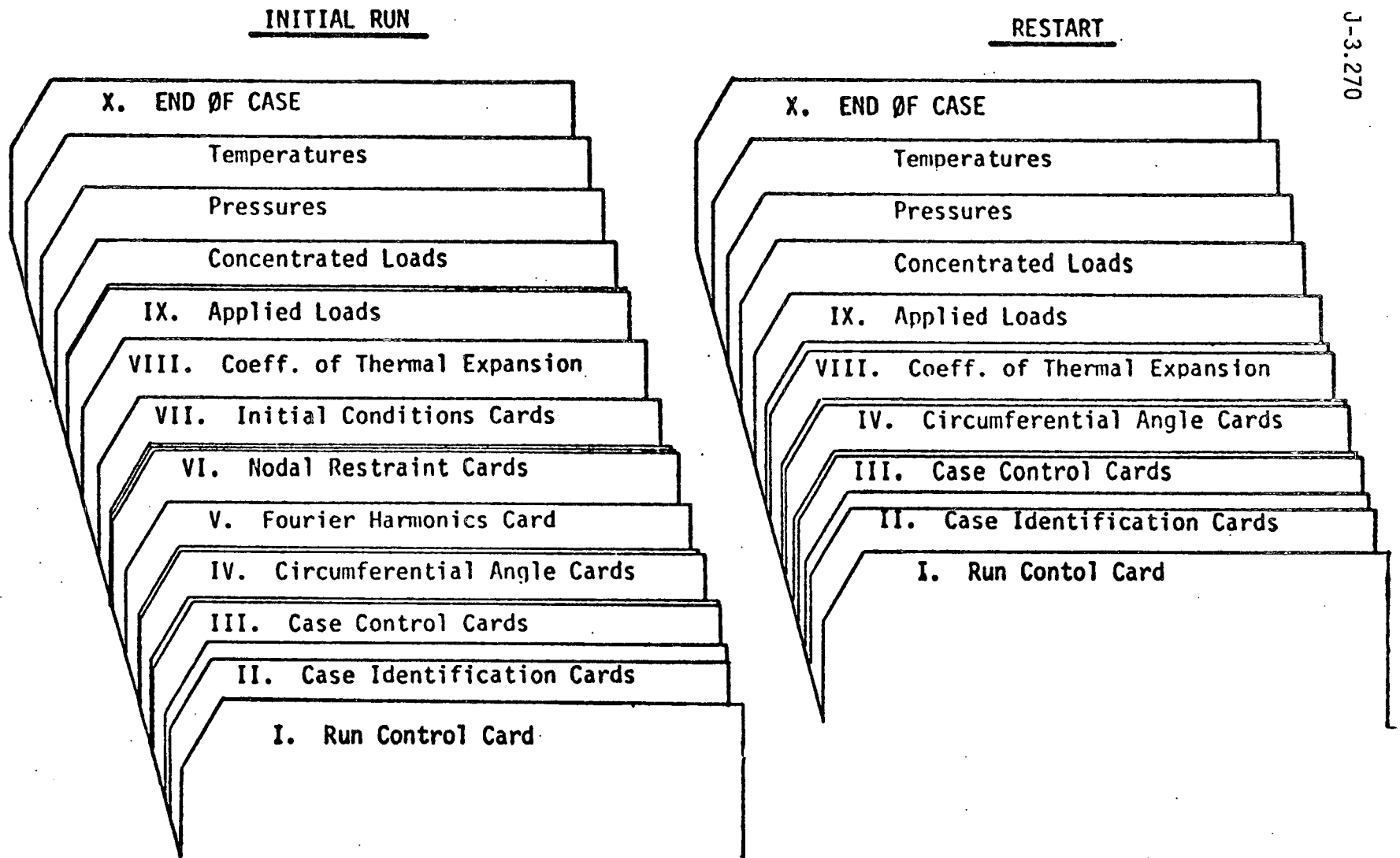


FIG. 1 CONSTITUTION OF DATA DECKS - INITIAL RUN AND RESTART MODES.

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## I. RUN CONTROL CARD

This card is used to identify the number of cases to be run and the logical unit numbers of the scratch tapes used in the run. (ONLY ONE CARD I IS USED PER RUN.)

Card Type I Format (3I5)		
Columns	Variable	Description
1-5	NCASES	The number of different data sets utilized for this run. .....
6-10	ND	Logical unit number of the scratch tape onto which all the data is read at the start of the run. .....
11-15	NS	Logical unit number of a second scratch tape used by the program. .....

## II. CASE IDENTIFICATION CARDS

These cards allow the user to print out comments which identify the problem being run.

A. Control Card (ONE CARD II-A PER DATA SET)

Columns	Variable	Description
1-5	NCARDS	Number of comment cards (TYPE II-B) which follow. .....
6-10	NT	Logical unit number of the tape (prepared by SAMMSOR) from which the stiffness and mass matrices, element properties, and re-start information, if needed, will be read. .....

B. Identification Cards - The information punched on these cards is printed as output and should identify the problem being run. These comments should not duplicate those of the SAMMSOR case since the SAMMSOR comments will also appear as output. (IF NCARDS=0, OMIT CARDS II-B, OTHERWISE INCLUDE NCARDS OF TYPE II-B.)

Columns	Variable	Description
1-80	COMENT	Any desired alphanumeric information may be printed on these cards. .....



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## III. CASE CONTROL CARDS

- A. Control Constants - Time parameters, restart information, and other miscellaneous control constants are input on this card. (INCLUDE ONE CARD III-A PER DATA SET.)

Card Type III-A Format (2F10.0,4I5)		
Columns	Variable	Description
1-10	TOTIME	The maximum time (seconds) for which the calculations are to be performed. .....
11-20	DELTE	Time increment (seconds) used in solving the equations of motion. .....
21-25	IRSTRT	Control constant which indicates if the solution is being restarted. If the solution is being restarted set IRSTRT = 1. If not, set IRSTRT = 0. .....
26-30	INCRST	The number of the time increment at which the solution is to be restarted. INCRST must be an integer multiple of the value of NPRNIT used in the previous run. If IRSTRT = 0, set INCRST = 0. .....
31-35	NCLOSE	For a closed shell (such as a spherical cap or a hemisphere) where node 1 is at the apex, set NCLOSE = 1. Radial and rotational restraints will then be applied for the zeroth harmonic to aid the numerical stability of the solution. If the shell does not fit the above description, set NCLOSE = 0. .....
36-40	ITELF	If thermal loads are to be applied in the program, set ITEL = 1. Otherwise, set ITEL = 0. .....

- B. Print Control Card - The constants used to control the program output are punched on this card. (INCLUDE ONE CARD III-B PER DATA

SET.)

Card Type III-B Format (10I5)		
Columns	Variable	Description
1-5	NPRNTQ	If the displacements are to be printed, set NPRNTQ = 1. If not, set NPRNTQ = 0. .....
6-10	IPRINT	If NPRNTQ = 1, the displacements will be printed every IPRINT time increments beginning with the first time step. If NPRNTQ = 0, set IPRINT = 0. .....
11-15	NCLCST	If the stresses and stress resultants are to be calculated, set NCLCST = 1. If not, set NCLCST = 0. .....
16-20	NSTRSS	If NCLCST = 1, the stress and stress resultants will be calculated and printed every NSTRSS time increments beginning with the first step. If NCLCST = 0, set NSTRSS = 0. .....
21-25	NPRNT	If restart information is to be placed on tape, set NPRNT = 1. If not, set NPRNT = 0. .....
26-30	NPRNIT	If NPRNT = 1, the restart information will be written on the output tape every NPRNIT time increments. If NPRNT = 0, set NPRNIT = 0. It is suggested that relatively large values of NPRNIT be used, say 200, 400, etc., if the total number of time steps is relatively large. .....
31-35	NPRNTL	If a printout of the applied loads is desired, set NPRNTL = 1. Otherwise, set NPRNTL = 0. .....
36-40	NPRNTF	If a printout of the generalized forces is desired, set NPRNTF = 1. Otherwise, set NPRNTF = 0. .....

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	41-45	NPRNTH	If the Fourier coefficients for the temperature and temperature gradient are to be printed, set NPRNTH = 1. Otherwise, set NPRNTH = 0. .....	
	46-50	NPRNMS	If the mass and stiffness matrices are to be printed, set NPRNMS = 1. If not, set NPRNMS = 0. .....	

IV. CIRCUMFERENTIAL ANGLE CARDS

The circumferential angles at which the displacements and stresses are to be calculated are read from these cards.

A. Control Card - (ONE CARD IV-A PER DATA SET.)

Card Type IV-A Format (I5)		
Columns	Variable	Description
1-5	NTHETA	The number of circumferential angles at which the displacements and possibly stresses are to be calculated. (1 ≤ NTHETA ≤ 20) .....

B. Circumferential Angles - (INCLUDE 1-3 CARDS IV-B PER DATA SET, DEPENDING UPON THE VALUE OF NTHETA.)

Card Type IV-B Format (8F10.0)		
Columns	Variable	Description
1-10	THETA(1)	Circumferential angles at which the displacements and possible stresses will be calculated. .....
11-20	THETA(2)	(If it is desired to calculate the displacements only along the line = 0, then include one card IV-B and set THETA(1) = 0.0)
"	"	
"	THETA(NTHETA)	

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V. FOURIER HARMONICS CARD

This card provides the number of Fourier cosine harmonics to be employed for this analysis and enumerates the specific harmonics to be used. (IF IRSTRT = 1, OMIT CARD V. OTHERWISE, INCLUDE ONE CARD V PER DATA SET.)

Card Type V Format (6I5)		
Columns	Variable	Description
1-5	NH	The total number of Fourier cosine harmonics to be utilized in this analysis ( $1 \leq NH \leq 5$ ). .....
6-10	IHARM(1)	Specific harmonics numbers to be employed. NH values must be given and the zero harmonic must always be specified as one of the input harmonic numbers. The user should check to be certain that the information for each of these harmonics has been created and stored on tape by the SAMMSOR code. .....
11-15	IHARM(2)	
16-20	IHARM(3)	
21-25	IHARM(4)	
26-30	IHARM(5)	

Example: Consider a case where it is desired to utilize harmonics 0, 2, 3, and 4. The input data for card V would then utilize the following values:

```

NH           = 4
IHARM(1)    = 0   NOTE:  IHARM(1) should always
                    be set equal to zero.
IHARM(2)    = 2
IHARM(3)    = 3
IHARM(4)    = 4
    
```

Columns 26-30 corresponding to IHARM(5) should be left blank for this example since only four harmonics are being run.

VI. NODAL RESTRAINT CARDS (Boundary Conditions)

The displacement constraints applied to the shell are described utilizing these cards. (IF IRSTRT = 1, OMIT CARDS VI-A AND VI-B.)

A. Control Card - (ONE CARD VI-A PER DATA SET, UNLESS IRSTRT = 1.)

Columns	Variable	Description
1-5	NODRES	Total number of displacement constraints to be applied to the shell ( $0 \leq \text{NODRES} \leq 204$ ) .....

B. Boundary Conditions - (THE NUMBER OF CARDS OF TYPE VI-B MUST EQUAL NODRES, UNLESS IRSTRT = 1. IF NODRES = 0, OMIT CARDS VI-B.)

Columns	Variable	Description
1-5	NP	Number of the node where the restraint is to be applied. .....
6-10	NDIRCT	Key used to indicate the degree of freedom which is restrained.  NDIRCT = 1 applies axial restraint NDIRCT = 2 applies circumferential restraint NDIRCT = 3 applies radial restraint NDIRCT = 4 applies rotational restraint .....

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## VII. INITIAL CONDITIONS CARDS

The initial velocities and displacements of the nodes are specified on these cards. (IF IRSTRT = 1, OMIT CARDS VII-Z, VII-B, AND VII-C.)

- A. Control Card - Utilization of this control card greatly simplifies the specification of the initial conditions if either the initial velocities or the initial displacements, or both, are equal to zero. (ONE CARD VII-A PER DATA SET)

Columns	Variable	Description
1-5	IQN	If the initial velocities at all the nodes are zero, set IQN = 0. If not, set IQN = 1. .....
6-10	IQN1	If the initial displacements at all the nodes are zero, set IQN1 = 0. If not, set IQN1 = 1. .....

- B. Initial Velocities - The initial nodal velocities must be specified for each node of the shell for each harmonic to be run. The logic used to input the nodal velocities is essentially the same as the procedure used to specify the element properties in the SAMMSOR code. The initial velocities for each of the nodes are specified for the first of the input harmonics, then for the second input harmonic, etc. This process is repeated until the nodal velocities for the first of the input harmonics, then for the second input harmonic, etc. This process is repeated until the nodal velocities for each harmonic have been specified. (IF IQN = 0, OMIT CARDS

VII-B.)

Card Type VII-B Format (2I5, 4F10.0)

Columns	Variable	Description
1-5	IN1	First node to which the velocities specifies on this card are applied. .....
6-10	IN2	Last node to which the velocities specified on this card are applied. .....
11-20	$\dot{q}_1$	Initial nodal velocity in the axial direction for a particular harmonic. .....
21-30	$\dot{q}_2$	Initial nodal velocity in the circumferential direction for a particular harmonic. .....
31-40	$\dot{q}_3$	Initial nodal velocity in the radial direction for a particular harmonic. .....
41-50	$\dot{q}_4$	Initial nodal rotational velocity in the meridional direction for a particular harmonic. .....

C. Initial Displacements - In identically the same manner as is utilized for the initial velocities, the initial displacements are



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specified for each harmonic. (IF IQN1 = 0, OMIT CARDS VII-C)

Card Type VII-C Format (2I5, 4F10.0)		
Columns	Variable	Description
1-5	IN1	First node to which the displacements specified on this card are applied. .....
6-10	IN2	Last node to which the displacements specified on this card are applied. .....
11-20	q 1	Initial nodal displacement in the axial direction for a particular harmonic. .....
21-30	q 2	Initial nodal displacement in the circumferential direction for a particular harmonic. .....
31-40	q 3	Initial nodal displacement in the radial direction for a particular harmonic. .....
41-50	q 4	Initial nodal rotation in the meridional direction for a particular harmonic. .....

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VIII. COEFFICIENTS OF THERMAL EXPANSION

If the thermal effects are to be included in the analysis, the coefficients of thermal expansion must be specified using these cards. These coefficients are assumed to be constant for a given element but may vary from element to element. These coefficients are read in the same manner as the element properties in the SAMMSOR code. (THE NUMBER OF CARDS VIII MUST BE  $\leq$  NELEMS FOR ANY GIVEN DATA SET. IF ITELF = 0, OMIT CARDS VIII.)

Card Type VIII Format (2I5, 2F10.0)		
Columns	Variable	Description
1-5	IELM1	Number of the first element to which the properties on this card apply. .....
6-10	IELM2	Number of the last element to which the properties on this card apply. .....
11-20	ALSI1	Coefficient of thermal expansion in the meridional direction (in/in/deg). .....
21-30	ALTI1	Coefficient of thermal expansion in the circumferential direction (in/in/deg). .....

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## IX. APPLIED LOADS, TEMPERATURES, AND TEMPERATURE GRADIENTS

Since the concentrated nodal loads, distributed pressures, temperatures, and temperature gradients may vary in time; it may be necessary to specify these loads at a number of points in time. If these loads and temperatures are input at times  $T_{1j}$  and  $T_{1j+1}$ , the program will calculate generalized forces due to these loads at each of the input times. A linear variation of the generalized forces is then assumed between the times the loads are input. As soon as the value of the time reaches  $T_{1j+1}$ , a new set of loads is read in at  $T_{1j+2}$  and the process of calculating the generalized forces is repeated. The time increment, DELTE (CARD III-A), used in the solution of the equations of motion must be less than the difference between any two of the times at which the loads are specified. If the loads and/or temperatures propagate in and direction (moving loads), it is advisable to specify the loads at more times than is necessary if they vary in intensity only.

Ring loads can be applied at the nodes and must be input for each of the harmonics. The ring loads utilize the same sign convention employed for the shell nodal displacements.

The pressure loadings, temperatures and temperature gradients are assumed constant over the meridional length of the element but variations in the circumferential direction are allowed. These loadings may be input in one of two ways. Either the Fourier coefficients can be specified for each harmonic or the values of the loads may be specified at a number of circumferential angles around the shell elements. Utilizing this second procedure a step function variation is assumed in the circumferential direction. That is, the load is assumed constant from  $\theta_j$  to  $\theta_{j+1}$  with the value of the loads being equal to those specified at  $\theta_j$ . Sign conventions for the pressure loading are given in Figure 2.

A control card (Card Type IX-A) containing several key variables is used to guide the reading of the loading conditions. Proper selection of the values of these key variables results in a highly efficient procedure for specifying a wide variety of loading conditions. The key words and their meanings are explained in Figure 3.

Before attempting to input loads to the code the user is advised to study the guidelines presented in Section II, the example problems of Section II, and Appendix 6 which presents a thorough discussion of the various procedures necessary for specifying the loads.

### A. Load Control Card

This control card is utilized to direct the input of the loads for a given time. This card indicates the presence or absence of concentrated

forces and distributed pressure loadings and indicates the procedure to be utilized for creating the generalized thermal forces. (ONE CARD IX-A IS NECESSARY FOR EACH TIME AT WHICH THE LOADS ARE BEING INPUT.)

Card Type IX-A Format (F10.0, 4I5, A8)		
Columns	Variable	Description
1-10	T1	The time for which the loads are being input (sec). .....
11-15	NCF	If concentrated ring loads are applied to the structure at time T1, set NCF = 1. If not, set NCF = 0. .....
16-20	IDELF	If distributed loads are to be applied to the shell at time T1, set IDELF = 1. If not, set IDELF = 0. .....
21-25	IDCOE	If the Fourier cosine coefficients for the distributed loadings are to be read in at time T1, set IDCOE = 1. If not, set IDCOE = 0. .....
26-30	ITCOE	If the Fourier cosine coefficients for the temperatures and temperature gradients are to be read in at time T1, set ITCOE = 1. If not, set ITCOE = 0. .....
31-38	CONSTF	If the applied loads, temperatures and temperature gradients are constant from time, T1, to the final time, TDFIME (CARD III-A), punch the word CONSTANT in columns 31-38. If these parameters are not constant, leave columns 31-38 blank. .....

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**B. Concentrated Ring Loads**

The concentrated ring loads must be specified for each harmonic. (IF NCF = 0, OMIT CARDS IX-B.)

1. Control Card - This card indicates the presence or absence of concentrated ring loads for a particular harmonic. (ONE CARD IX-B-1 FOR EACH HARMONIC.)

Card Type IX-B-1 Format (I5)		
Columns	Variable	Description
1-5	NCF1	If there are concentrated ring loads for this particular harmonic, set NCF1 = 1. If not, set NCF1 = 0. .....

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2. Concentrated Ring Loads - For harmonics having ring loads associated with them, the loads are specified using these cards. (IF NCF1 = 0, OMIT CARDS IX-B-2 FOR THE HARMONIC BEING CONSIDERED.) ONE OR MORE CARDS IX-B-2 MAY BE USED, BUT NEVER UTILIZE MORE THAN 51 PER HARMONIC.

Card Type IX-B-2 Format (2I5, 4F10.0)		
Columns	Variable	Description
1-5	IN1	First node to which this loading applies. .....
6-10	IN2	Last node to which this loading applies. .....
11-20	F1	Axial ring load applied at a node (lb).* .....
21-30	F2	Circumferential ring load applied at a node (lb).* .....
31-40	F3	Radial ring load applied at a node (lb).* .....
41-50	F4	Concentrated moment applied at a node (in-lb).* .....

Examples: The use of cards IX-B should become clear after considering the following examples:

1. Consider the case where a uniform tensile ring loading of 100 psi is being applied in the axial direction to the first node of a cylinder. The solution for this problem has been presented in Figure 20 of Reference 31 The thickness of the cylinder is 0.1

\* The total value of the ring load for each harmonic is input, not the load per unit length of circumference. For complicated ring loads the value of the load input for each harmonic is obtained by intergrating the product of the load and the corresponding displacement function around the circumference.

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inches with the radius being given as 6 inches. Consider that harmonics 0 and 2 are being run. The total ring load for the zero harmonic will be  $(100) \times 2\pi(6) \times (0.1) = 376.9$  lb.

Five cards of type IX are required to input these loads assuming they are constant from time  $T1 = 0.0$  to  $TOTIME$  and assuming 50 elements are used to idealize the structure.

CARD	VARIABLE	VALUES
IX-A	T1 = 0.0	NCF = 1                      IDELF = IDCOE = ITCOE = 0
IX-B	NCF1 = 1	(HARMONIC 0)
IX-C	IN1=1 IN1=1	F1 = -376.9    F2 = F3 = F4 = 0
IX-C	IN1 = 2    IN1 = 51	F1 = F2 = F3 = F4 = 0
IX-B	NCF2 = 0	(HARMONIC 2)

2. The second example considers a radial ring load of  $F \cos\theta$  applied to a cylinder of radius  $r$ .

Performing the integration, one obtains the radial ring load for harmonic 1 as

$$F_3 = \int_0^{2\pi} (F \cos\theta) r \cos\theta d\theta$$

$$= \pi r F$$

The Fourier coefficients for the other harmonics are zero.

### C. Distributed Loads - (IF IDELF = 0, OMIT CARDS IX-C)

The distributed loadings may be input in one of two ways: the Fourier coefficients may be read in for each harmonic or the loadings may be specified at a desired number of circumferential angles ( $\leq 37$ ). If the second option is used, the Fourier coefficients will then be generated internally. The user should note that it is possible to input distributed loads in only one of two ways.

1. Distributed Loads - (Input at various circumferential angles)  
 Since the choice of the displacement functions utilized in this analysis necessitate the presence of loads symmetric about the meridian  $\theta = 0$ , it is necessary to specify the distributed loadings for angles from  $0^\circ \rightarrow 180^\circ$ . The code then assumes that the distribution from  $180^\circ \rightarrow 360^\circ$  is the mirror image of the input distribution. (IF IDCOE = 1, OMIT CARDS IX-C-1)

a. Control Card - Utilize this card to indicate the number of angles for which the loads will be specified.

Card Type IX-C-1-a Format (3I5)		
Columns	Variable	Description
1-5	IELM1	First element to be distributed loading applies. .....
6-10	IELM2	Last element to which this distributed loading applies. .....
11-15	NDP	Number of circumferential angles at which the distributed loads are to be specified ( $1 \leq NDP \leq 37$ ). If the loadings are constant in the circumferential direction set NDP = 1. .....

b. Distributed Loads at Specified Angles\* This card specifies the angle at which the loads are being input and provides the values of the loads at that angle. (INCLUDE NDP CARDS OF

\* The first loading must always be given for  $\theta = 0^\circ$ . The next loading is given at the angle where the load changes in value. If the load is constant with respect to  $\theta$ , only one card will be necessary to input the load. Do not input values for the loads at  $\theta = 180^\circ$  since the load at that angle will be equal in all cases to the load input at the previous value of THETAB.



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TYPE IX-C-1-b FOR EACH CARD IX-C-1-a.)

Card Type IX-C-1-b Format (4F10.0)		
Columns	Variable	Description
1-10	THETB	Circumferential angle (degrees) for which this data is given. .....
11-20	P	Distributed load in the meridional direction (psi). .....
21-30	R	Distributed load in the normal direction (psi). .....
31-40	S	Distributed load in the circumferential direction (psi). .....

Example: Consider the normal pressure distribution on an element depicted in Figure 4. To input the pressure on this element requires specification of the pressures for four values of  $\theta$ .

THETB	R (I)
0.0	-Q1
30.0	-Q2
90.0	-Q3
2.0	0.0

2. Distributed Loads - (Fourier Coefficients) The Fourier coefficients for the distributed loads may be specified using these cards. The coefficients must be specified (even though they may be zero) for each harmonic being employed in the analysis. The coefficients are specified for each harmonic of the first group of elements, then for each harmonic of the second group, etc. until the values have been input for all the elements. (IF IDCOE = 0, OMIT CARDS IX-C-2)

a. Control Card

Card Type IX-C-2-a Format (2I5)		
Columns	Variable	Description
1-5	IELN1	First element to which these loads apply. .....
6-10	IELN2	Last element to which these loads apply. .....

b. Fourier Coefficients - (NH CARDS OF TYPE IX-C-2-b FOR EACH CARD IX-C-2-a.)

Card Type IX-C-2-b Format (3F10.0)		
Columns	Variable	Description
1-10	P	Fourier coefficient of the distributed load in the meridional direction for a particular harmonic (psi). .....
11-20	R	Fourier coefficient of the distributed load in the normal direction for a particular harmonic (psi). .....
21-30	S	Fourier coefficient of the distributed load in the circumferential direction for a particular harmonic (psi). .....

D. Temperature Distribution and Gradients

Essentially the same logic is employed for inputting the temperatures and gradients that was used for the specification of the distributed loads. The explanation of this procedure should therefore not need be repeated.

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The temperatures are specified for the midsurface of the shell. The temperature gradients (through the thickness) are considered positive if the temperature for the outer surface is greater than the temperature on the inner surface. (IF ITELF = 0, OMIT CARDS IX-D.)

1. Temperature Distribution and Gradients - (Input at various circumferential angles)

Again, the requirement of symmetry about the meridian  $\theta = 0$ , makes it necessary to specify the temperature distribution and thermal gradients only from  $0^\circ \rightarrow 180^\circ$ . The temperature distribution and gradients are input on the same cards for the various angles. (IF ITCOE = 1, OMIT CARDS IX-D-1.)

- a. Control Card - Utilize this card to indicate the number of angles for which the temperature and gradients will be specified.

Card Type IX-D-1-a Format (3I5)		
Columns	Variable	Description
1-5	IELM1	First element to which this data applies. .....
6-10	IELM2	Last element to which this data applies. .....
11-15	NDP	Number of circumferential angles at which the temperature distribution and gradient are to BE SPECIFIED ( $1 \leq NDP \leq 37$ ). If the temperature is constant in the circumferential direction, set NDP = 1. .....

b. Temperature and Temperature Gradient at Specified Angles -

This card specifies the angle at which the temperature and temperature gradient (through the thickness) is being input and provides the value of the temperature at that angle. (INCLUDE NDP CARDS OF TYPE IX-D-1b

FOR EACH CARD IX-D-1-a.)

Card Type IX-D-1-b Format (3F10.0)		
Columns	Variable	Description
1-10	THETB	Circumferential angle for which this temperature and gradient are given. .....
11-20	P	Distributed temperature at $\theta = \text{THETB}$ ( $^{\circ}\text{F}$ ). .....
21-30	R	Temperature gradient (through the thickness) at $\theta = \text{THETB}$ ( $^{\circ}\text{F}/\text{in}$ ). .....

2. Temperature Distribution and Gradient - (Fourier Coefficients)

If the user so desires, the Fourier coefficients for the temperature distribution and gradient may be specified for each of the harmonics being used. Again, the coefficients are specified for all harmonics for the first group of elements, then for the second group, etc., until all the element coefficients have been input. (IF ITCOE = 0, OMIT CARDS IX-D-2)

a. Control Card

Card Type IX-D-2-a Format (2I5)		
Columns	Variable	Description
1-5	IELM1	First element to which these properties apply. .....
6-10	IELM2	Last element to which these properties apply. .....

b. Fourier Coefficients - (NH CARDS OF TYPE IX-D-2-b FOR EACH

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CARD IX-D-2-a.)

Card Type IX-D-2-b Format (2F10.0)		
Columns	Variable	Description
1-10	TH1	Fourier coefficient of the temperature distribution (°F) for a particular harmonic. .....
11-20	DTH1	Fourier coefficient of the temperature gradient (°F/in) for a particular harmonic. .....

X. FINAL DATA CARD FOR A CASE

Place this card after the last card IX of each data set. This signifies the end of the input data for a case. (ONE CARD X PER DATA SET.)

Card Type X	
Columns	Punch
1-11	END OF CASE .....

XI. FINAL DATA CARD FOR A RUN

This card must be placed after the card X of the last case to be run.

It denotes the end of the input data for a run. (ONE CARD XI PER RUN)

Card Type XI	
Columns	Punch
1-10	END OF RUN .....

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### EXAMPLE PROBLEMS

The example problems which follow were chosen to demonstrate the versatility of the code and to further acquaint the users with the procedures for inputting the data to the code. The data presented herein is typical for the problems solved by the code and demonstrates many of the input procedures.

Since the most complex portion of the input data is the specification of the loading conditions, a variety of loadings are demonstrated. Response curves are presented so the user may check his output with the previously obtained curves. The first two example problems utilize the shells described in example problems 1 and 2 of the SAMMSOR user's guide (Ref. 1) while the third example problem demonstrates the two procedures for specifying distributed pressure loadings.

#### Example Problem 1

The first example problem was chosen to demonstrate the procedure for inputting a concentrated ring load and to demonstrate the program's capability to solve highly nonlinear problems. For the forty pound load applied in this problem, the static solution shows that the nonlinear displacement is more than four times as large as the linear solution.

The shell to which the load is applied is the shallow spherical cap ( $\lambda=6$ ) utilized in the first example problem in the SAMMSOR user's guide. The edges of the shell are assumed to be clamped. Since the loading is symmetric, the displacements and stresses will be calculated only along the line  $\theta = 0$ . Only the response for the zeroth harmonic will be determined. A set of input data for this case is presented in Figure 5 with the displacement response of the apex of the shell being presented in Figure 6. This response curve should allow the user to check his version of the code.

#### Example Problem 2

The shell described in the second example problem in the SAMMSOR user's guide is now subjected to a 50 psi internal pressure. The load-in is applied at time  $T1 = 0.0$  and remains constant for the duration of the calculation.

Two sets of input data are provided for this example problem. The first set (Figure 7) allows the program to calculate the response for the first 300 time steps. The second set of input data (Figure 8) will

NCASE= 1

PRINTOUT OF INPUT DATA

CARD TYPE	10	20	30	40	50	60	70	80
II - A	6	4						
- B	*****							
- B	EXAMPLE PROBLEM NO. 1				DYNASOR II USER'S MANUAL			
- B	THE SHELL DESCRIBED IN EXAMPLE PROBLEM 1 OF THE SAMMSOR USER'S GUIDE							
- B	IS SUBJECTED TO A 40 LB. APEX LOADING WITH THE SOLUTION BEING DETERMINED							
- B	FOR 400 TIME STEPS							
- B	*****							
III - A	0.0001	.00000025	0	0	1	0		
- B	1	4	1	8	1	100	1	0 1
IV - A	1							
- B		0.0						
V	1	0						
VI - A	4							
- B	31	1						
- B	31	2						
- B	31	3						
- B	31	4						
VII - A	0	0						
IX - A		0.0	1	0	0	OCONSTANT		
- B	- 1	1						
- B	- 2	1	1	40.0	0.0	0.0	0.0	0.0
- B	- 2	2	31	0.0	0.0	0.0	0.0	0.0
X	END OF CASE							

Fig. 5 INPUT DATA - EXAMPLE PROBLEM 1



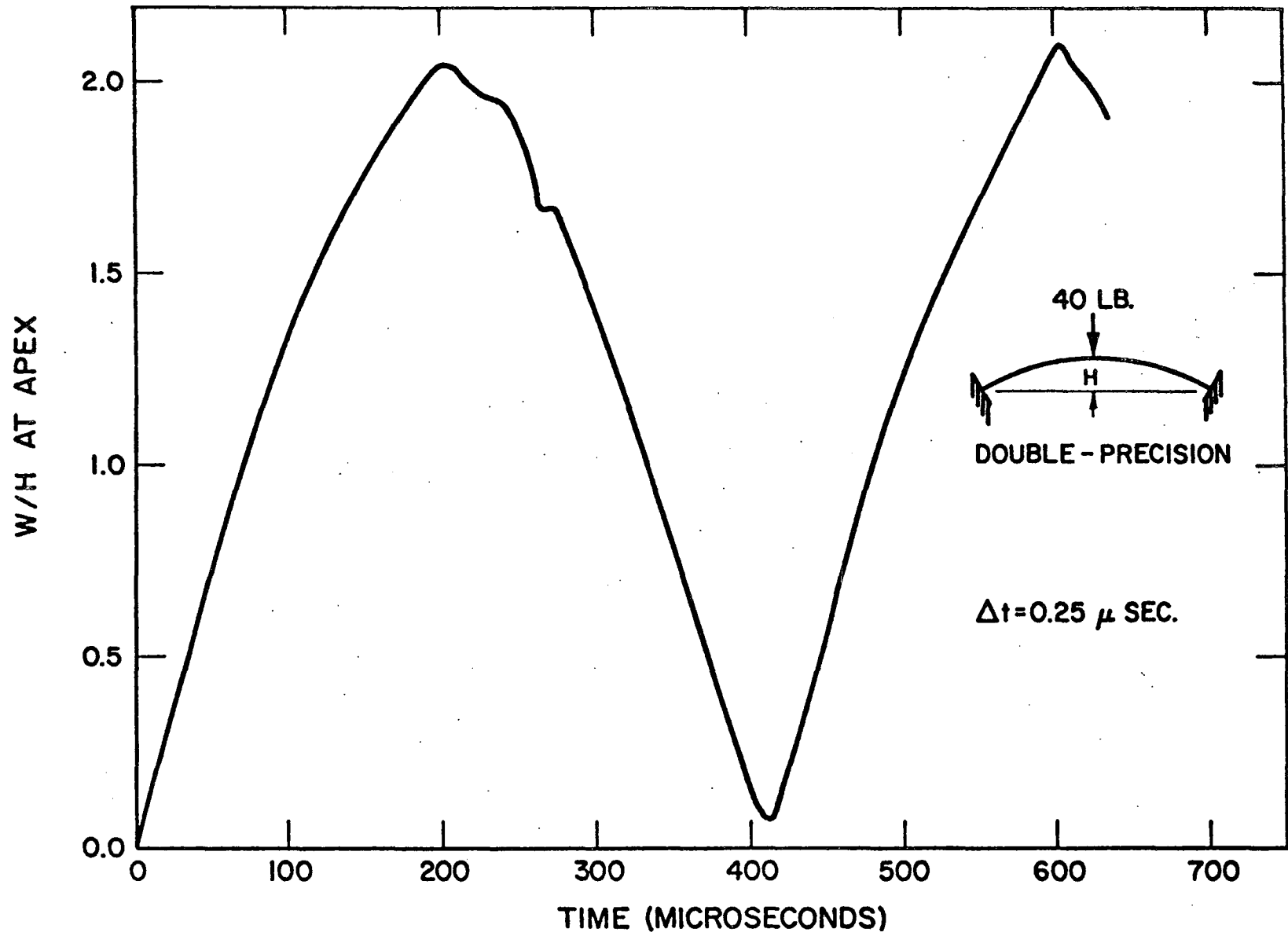


FIG. 6 APEX DISPLACEMENT RESPONSE UNDER CONCENTRATED AXIAL LOAD

NCASE= 2

PRINTOUT OF INPUT DATA

Fig. 7 INPUT DATA - EXAMPLE PROBLEM 2

CARD	10	20	30	40	50	60	70	80
TYPE	12345678901	2345678901	2345678901	2345678901	2345678901	2345678901	2345678901	234567890
II - A	6	8						
- B	*****							
- B	EXAMPLE PROBLEM NO. 2				DYNASOR II USER'S MANUAL			
- B	CAP-TORUS-CYLINDER CONFIGURATION							
- B	THE SHELL DEPICTED IN THE SECOND EXAMPLE PROBLEM OF THE SAMMSOR USER'S							
- B	MANUAL IS SUBJECTED TO A 50 PSI INTERNAL PRESSURE							
- B	*****							
III - A	0.0009	0.000003	0	0	1	0		
- B	1	10	1	20	1	100	1	0 1
IV - A	1							
- B		0.0						
V	1	0						
VI - A	4							
- B	51	1						
- B	51	2						
- B	51	3						
- B	51	4						
VII - A	0	0						
IX - A	0.0	0	1	0	OCONSTANT			
- C	-1-a 1	50	1					
	-b	0.0	0.0	50.0	0.0			
X	END OF CASE							

Fig. 8 INPUT DATA - EXAMPLE PROBLEM 2 - RESTART MODE

NCASE= 3

PRINTOUT OF INPUT DATA

CARD TYPE	10	20	30	40	50	60	70	80
	12345678901	2345678901	2345678901	2345678901	2345678901	2345678901	2345678901	2345678901
II - A	6	8						
- B	*****							
- B	EXAMPLE PROBLEM NO. 2				DYNASOR II USER'S MANUAL			
- B	THE INPUT DATA NECESSARY TO RESTART THE CODE AT TIME INCREMENT 300							
- B	IS PROVIDED TO GUIDE THE USER IN HIS RESTART OPERATIONS. THE PROBLEM							
- B	IS TO BE RUN FOR AN ADDITIONAL 300 TIME INCREMENTS.							
- B	*****							
III - A	0.0018	0.000003	1	300	1	0		
- B	1	10	1	20	1	1	0	0
IV - A	1							
- B	0.0							
IX - A	0.0009	0	1	0	OCONSTANT			
- C	- 1 -a	1	50	1				
	-b	0.0	0.0	50.0	0.0			
X	END OF CASE							

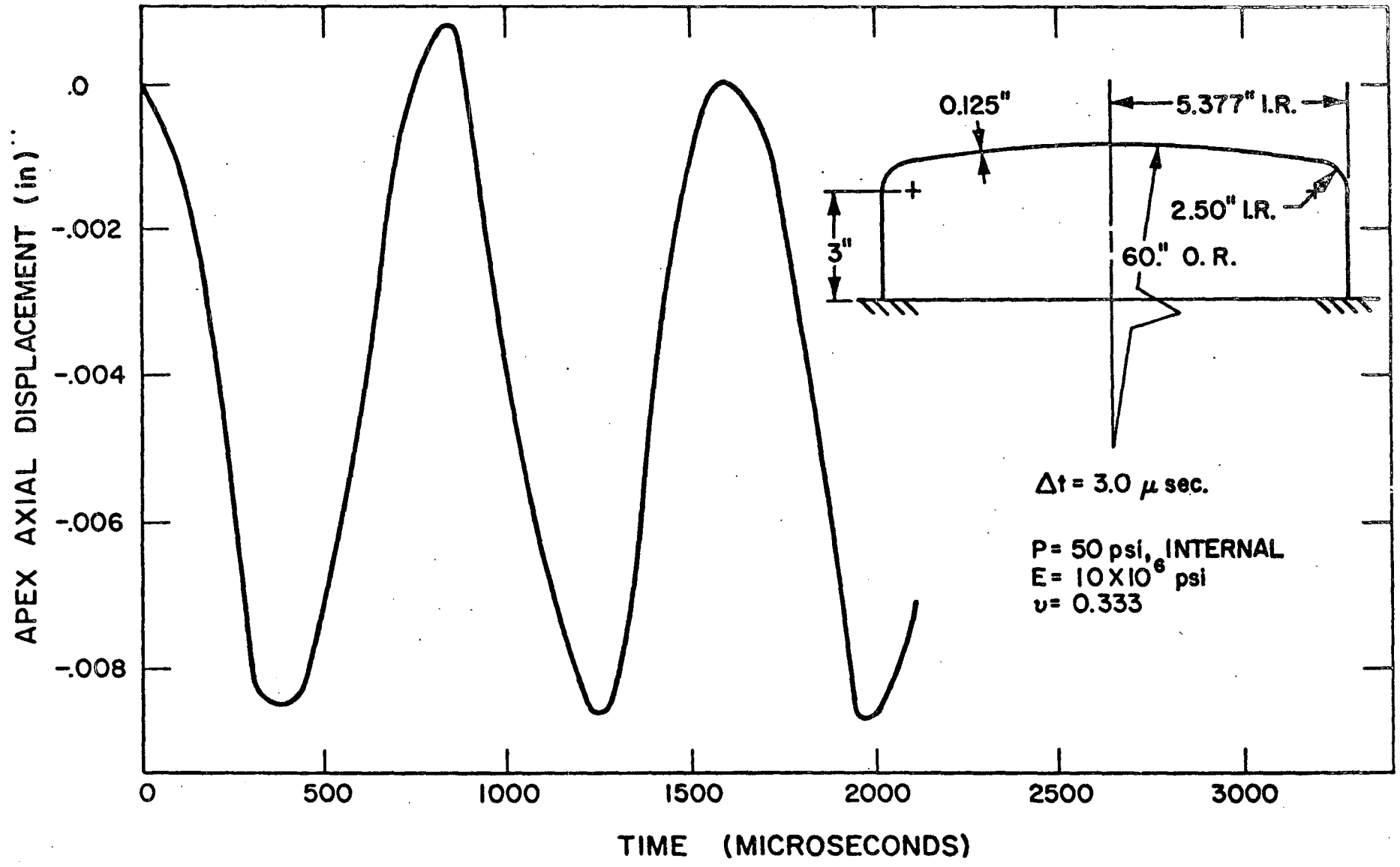


FIG. 9 DISPLACEMENT RESPONSE UNDER INTERNAL PRESSURE

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restart the code at the end of the 300th time step and will then allow the program to calculate the response for an additional 300 increments.

Since this problem is only moderately nonlinear, it is interesting to note that a much larger time step can be used for this problem than was employed in the previous example problem. The displacement response obtained for this problem is presented in Figure 9.

### Example Problem 3

This example problem was selected to demonstrate the procedures for inputting the distributed loadings on a shell. A cylindrical shell (figure 10) is subjected to a half cosine loading which is symmetric about the meridian  $= 0$ . This load is applied along the entire length of the shell. The pressure loading may be specified in one of two ways:

- 1) The Fourier coefficients may be input for each harmonic.
- 2) The pressure may be specified at various circumferential angles with the Fourier coefficients then being internally generated.

The first set of input data (Figure 11) utilizes the first of the above procedures and inputs the Fourier coefficients. The input data presented in Figure 12 describes the loading by specifying the value of the pressure at the various angles. The same procedure is employed to describe the temperature and temperature gradient distributions.

Considering the symmetry of the loading and the boundary conditions applied to this shell, it can easily be recognized that the displacements and stresses will be symmetric about the center of this cylindrical tube. Therefore, only one-half of the shell needs to be analyzed. The plane of symmetry is assured by applying an axial and a rotational restraint at node one (1).

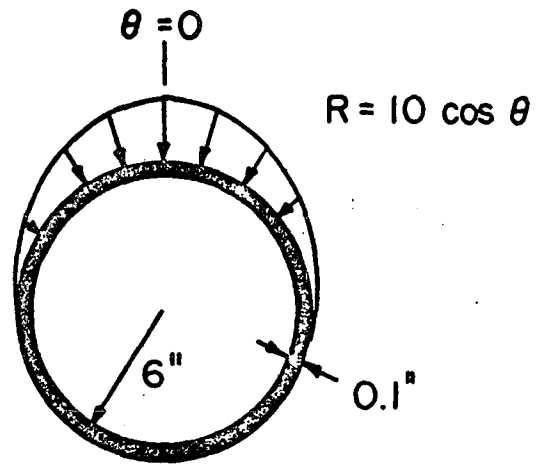
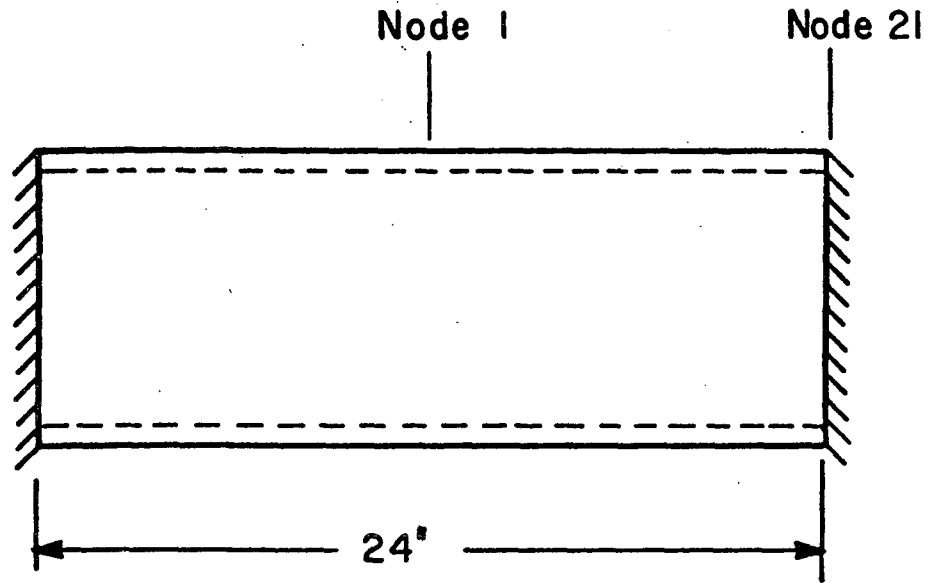


FIG 10 CYLINDRICAL SHELL SUBJECTED TO HALF COSINE PRESSURE LOADING

NCASE= 5

PRINTOUT OF INPUT DATA

Fig. 11 INPUT DATA - (SET #1) - EXAMPLE PROBLEM 3

CARD TYPE	10	20	30	40	50	60	70	80
	12345678901	2345678901	2345678901	2345678901	2345678901	2345678901	2345678901	2345678901
II - A	6	4						
- B	*****							
- B	EXAMPLE PROBLEM NO. 3				DYNASOR II USER'S MANUAL			
- B	CYLINDRICAL SHELL IDEALIZED USING 30 ELEMENTS IS SUBJECTED TO A HALF COSINE							
- B	LOADING TO DEMONSTRATE THE OPTIONS FOR INPUTTING DISTRIBUTED LOADS.							
- B	** IN THIS CASE THE PRESSURE IS SPECIFIED BY INPUTTING THE FOURIER COEFFICIENTS							
- B	*****							
III - A	0.0005	0.00001	0	0	0	0		
- B	1	5	1	10	1	50	1	1 0 1
IV - A	2							
- B		0.0		30.0				
V - A	5	0	1	2	3	4		
VI - A	6							
- B	1	1						
- B	1	4						
- B	21	1						
- B	21	2						
- B	21	3						
- B	21	4						
VII - A	0	0						
IX - A		0.0	0	1	1			0CONSTANT
- C	- 2-	a1	20					
- b		0.0		-3.1831				0.0
- b		0.0		-5.0000				0.0
- b		0.0		-2.1221				0.0
- b		0.0		0.0000				0.0
- b		0.0		0.4244				0.0
X	END OF CASE							

PRINTOUT OF INPUT DATA

CARD TYPE	10	20	30	40	50	60	70	80
	12345678901	23456789012	34567890123	45678901234	56789012345	67890123456	78901234567	8901234567890
II - A	6	4						
- B	*****							
- B	EXAMPLE PROBLEM NO. 3				DYNASOR II USER'S MANUAL			
- B	CYLINDRICAL SHELL IDEALIZED USING 30 ELEMENTS IS SUBJECTED TO A HALF COSINE							
- B	LOADING TO DEMONSTRATE THE OPTIONS FOR INPUTTING DISTRIBUTED LOADS.							
- B	** IN THIS CASE THE PRESSURE IS SPECIFIED AT VARIOUS CIRCUMFERENTIAL ANGLES **							
- B	*****							
III - A	0.0005	0.00001	0	0	0	0		
- B	1	5	1	10	1	50	1	1 0 0
IV - A	2							
- B		0.0	30.0					
V	5	0	1	2	3	4		
VI - A	6							
- B	1	1						
- B	1	4						
- B	21	1						
- B	21	2						
- B	21	3						
- B	21	4						
VII - A	0	0						
IX - A		0.0	0	1	0	OCONSTANT		
- C - 1 - a	1	0.0	37					
- b		0.0		0.0	- 9.9976			0.0
- b		2.5		0.0	- 9.9786			0.0
- b		5.0		0.0	- 9.9406			0.0
- b		7.5		0.0	- 9.8836			0.0
- b		10.0		0.0	- 9.8079			0.0
- b		12.5		0.0	- 9.7134			0.0
- b		15.0		0.0	- 9.6005			0.0
- b		17.5		0.0	- 9.4693			0.0
- b		20.0		0.0	- 9.3201			0.0
- b		22.5		0.0	- 9.1531			0.0
- b		25.0		0.0	- 8.9687			0.0
- b		27.5		0.0	- 8.7673			0.0
- b		30.0		0.0	- 8.5491			0.0
- b		32.5		0.0	- 8.3147			0.0
- b		35.0		0.0	- 8.0644			0.0
- b		37.5		0.0	- 7.7988			0.0
- b		40.0		0.0	- 7.5184			0.0
- b		42.5		0.0	- 7.2236			0.0
- b		45.0		0.0	- 6.9151			0.0
- b		47.5		0.0	- 6.5935			0.0
- b		50.0		0.0	- 6.2592			0.0
- b		52.5		0.0	- 5.9131			0.0
- b		55.0		0.0	- 5.5557			0.0
- b		57.5		0.0	- 5.1877			0.0
- b		60.0		0.0	- 4.8059			0.0
- b		62.5		0.0	- 4.4229			0.0
- b		65.0		0.0	- 4.0275			0.0
- b		67.5		0.0	- 3.6244			0.0
- b		70.0		0.0	- 3.2144			0.0
- b		72.5		0.0	- 2.7983			0.0
- b		75.0		0.0	- 2.3769			0.0
- b		77.5		0.0	- 1.9509			0.0
- b		80.0		0.0	- 1.5212			0.0
- b		82.5		0.0	- 1.0887			0.0
- b		85.0		0.0	- 0.6540			0.0
- b		87.5		0.0	- 0.2181			0.0
- b		90.0		0.0	0.0000			0.0
X	END OF CASE							

Fig. 12 INPUT DATA - (SET #2) - EXAMPLE PROBLEM 3



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### Use of the Restart Option

In order for efficient use to be made of the DYNASOR II code, the user should become familiar with the option provided for restarting the program. Through effective use of this option the dynamic response studies can be completed using a minimum amount of computer time.

Use of the restart option may prove invaluable in a number of situations. Abnormal termination of the program may occur if a numerical instability is noted in the response. If this occurs, the restart option can be used with a different value of the time increment. Another important use of the restart option arises when the user is satisfied with the results previously obtained but desires to extend the response data to a further point in time. In such a case the program is restarted at the last time step for which the restart information was placed on tape. A most effective use of this option can be made when conducting dynamic stability analyses where it is desirable to evaluate the response to see if buckling has occurred. If it has not, the decision can then be made to extend the run to further points in time.

Utilizing large time steps can result in a damping effect upon the solution so it is advisable to run the problem for a couple of oscillations, check to see if the solution is significantly damped, and then run the problem for the desired number of oscillations. If an evaluation of the initial results indicates that a smaller or larger time step should be used, the restart facility might be used to keep from having to repeat the initial calculations.

The displacements, velocities, and forces should be written on tape for almost all of the cases to insure that the restart information will be available if an evaluation of the calculated response indicates that the program should be restarted. The time required to write the restart information on tape is negligible when compared with the amount of time required to obtain the total response.

If it is desirable to decrease the time increment when restarting the program, the user should exercise care in selection the increment (INRST) at which the program will be restarted. The decision to decrease the size of the time step will usually be based upon the observation that the solution has become unstable or that significant damping is present in the response. To restart the program the user must be sure that the increment (INCRST) has been selected small enough to insure that the inaccuracies created by the larger time step can be neglected.

On the other hand, if the results from a previous run indicate that it is possible to increase the size of the time step for the remaining calculations, then care must also be taken in the selection of INCRST. For the numerical extrapolation procedure to produce accurate sets of displacements, it is recommended that the solution be restarted on a

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relatively straight portion of the displacement response curve. Considering the curve presented in Figure 6, it would be recommended that the program be restarted at 500 microseconds rather than at 600 microseconds because of the extrapolation procedure being utilized (i.e. the curve is smoother at 500 microseconds).

When using the restart option, it is possible to specify different values for a number of the control constants and input parameters. The data on cards I-IV may be changed, but the same Fourier harmonics and boundary conditions must be used. It is also required that the coefficients of thermal expansion remain the same when restarting the program. These requirements allow the user to omit card types V, VI, and VII when preparing data for restart operations. The considerations effecting the input of the loads for restart operations are presented in Appendix 6.