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## MASS QUANTITY GAUGING BY RF MODE ANALYSIS

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Boulder, Colorado 80302

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Interim Report

Prepared for  
National Aeronautics and Space Administration  
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U.S. DEPARTMENT OF COMMERCE, Frederick B. Dent, Secretary  
NATIONAL BUREAU OF STANDARDS, Richard W. Roberts, Director

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## FOREWORD

Background - The RF technique is a gauging method which samples all parts of the inside of a tank. The response is characteristic of the total mass of fluid within a tank; this is one of the advantages of the RF system over the standard capacitance system which is essentially a local measurement of fluid density. Another advantage of the RF system is the simplicity of the hardware involved; a small grounded antenna about the size of a paper clip is sufficient to communicate with the inside of the tank.

Preliminary theoretical and experimental results on the RF gauging idea were obtained by NBS in connection with a NASA sponsored contract on slush hydrogen gauging. These results may be found in NBS Report 9793 dated June 1, 1971, on "Instrumentation for Hydrogen Slush Storage Containers."

Purpose - The purpose of this report is to summarize work done under purchase order T-1738B from the NASA Johnson Space Center Houston, Texas, to the National Bureau of Standards, Cryogenics Division, Boulder, Colorado. Items covered include:

- 1) Phase I - Preliminary studies of the radio frequency (RF) mass quantity gauging system for two phase and supercritical fluids; and construction of experimental system for detailed feasibility studies.
- 2) Phase II - Experimental evaluation of the system for supercritical nitrogen and hydrogen. (Oxygen is also included in Phase II; the results will be reported separately upon the completion of the oxygen testing.)

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Objective - The primary objective of this work is to design and develop a breadboard system to verify that the radio frequency resonant cavity mode analysis technique is conceptually sound for the fluid mass quantity gauging of the Space Shuttle Orbiter PRSD (Power Reactant Storage and Distribution) subsystem tankage, i. e., supercritically stored hydrogen and oxygen, in all gravity fields. The secondary program objective is to analytically determine the applicability of the concept to the quantity gauging of Shuttle Orbiter propulsion systems tankage. (i. e. sub-critical fluids).

End Product - The end product of this contractual effort is to be a breadboard RF mass quantity gauging system capable of gauging supercritically stored hydrogen and oxygen to an accuracy of one percent of total tank quantity in any gravity environment. This should also include the gauging of subcritical hydrogen and oxygen representative of fill and drain operations on the PRSD tanks.

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### ABSTRACT

This is a summary report of work done to date on NASA (Johnson Space Center) purchase order T-1738B concerning Radio Frequency (RF) Mass Quantity Gauging. Experimental apparatus has been designed and tested which measures the resonant frequencies of a tank in the "time domain." These frequencies correspond to the total mass of fluid within the tank. Experimental results are discussed for nitrogen and hydrogen in normal gravity both in the supercritical state and also in the two phase (liquid-gas) region. Theoretical discussions for more general cases are given.

Key Words: Gauging; hydrogen; nitrogen; radio frequency; total mass.

## INTRODUCTION

When a small antenna is placed in a closed metal cavity, the electromagnetic field pattern which the antenna generates inside the cavity depends on the excitation frequency and the shape of the cavity. At certain frequencies,  $f_n$  (called resonant frequencies), the field patterns are standing waves (called resonant modes). These modes are easily detected at the antenna since the impedance match of the antenna to the cavity is more efficient at the resonant frequencies.

The presence of a dielectric fluid within the cavity will change the resonant frequencies. The resonant frequencies will decrease with an increasing amount of fluid because the presence of the fluid slows down the propagation of the electromagnetic wave. This presents the possibility of gauging the amount of fluid by measuring the resonant frequencies.

If the density of fluid is uniform throughout the cavity,

$$f_n = \frac{f_{on}}{\sqrt{\epsilon}} \quad (1)$$

where  $f_{on}$  is the resonant frequency of the empty cavity for the  $n$ th mode and  $\epsilon$  is the dielectric constant of the fluid. For many nonpolar fluids of interest (including hydrogen and oxygen), the dielectric constant depends only on the density of the fluid. In this case there is a unique relationship between each resonant frequency and the total mass within the cavity; and only one mode is necessary to determine the total mass.

If the density of the fluid is not uniform throughout the cavity (either because of a two phase liquid-gas interface or a single phase fluid with temperature gradients) the resonant frequencies depend on the amount of fluid mass in the cavity and also somewhat on where the dense portions of the fluid are located within the cavity (fluid geometry effects). Since the geometry of each standing wave is different (the modes are linearly independent functions of the space variables), it is possible to partially compensate for the fluid geometry effects by comparing two or more modes. This process of comparison is called Mass Quantity Gauging by RF Mode Analysis.

The purpose of this present work is to develop an experimental system which will

- (1) Provide a breadboard total mass gauge for measuring uniform density fluids; developing the accuracy and data reduction of the time domain technique for measuring a single resonant frequency, and
- (2) Determine the feasibility of RF Mode Analysis for nonuniform fluids; to measure the total mass with sufficient accuracy for dynamic Normal-g and Zero-g fluid geometries.

This report will emphasize item (1) above under the heading Uniform Density Fluids and also give some preliminary discussion and results of item (2) above under the heading Nonuniform Density Fluids.



## EXPERIMENTAL SYSTEM

The experimental system consists of (1) an electronic signal conditioner and data acquisition system which measures the resonant frequencies in the time domain and (2) an 18 inch diameter spherical cryogenic storage tank (experimental vessel) designed for testing nitrogen and hydrogen, in both the two-phase and supercritical states; it is expected that after further testing that the tank will also be suitable for oxygen. The total fluid mass is determined by weighing with a calibrated load cell; there are a number of thermocouples and resistance thermometers attached to the sphere to measure temperature gradients in the fluid. There are several antenna locations for measuring the effect of antenna orientation with respect to a non-homogeneous fluid. The details of the cryogenic system are outlined in Appendix A.

The resonant frequencies may be detected by sweeping the antenna with an RF sweep generator; where the generator frequency ranges between  $f_A$  and  $f_B$  and the resonant frequency (or frequencies) of interest lies between  $f_A$  and  $f_B$ . When the generator frequency coincides with the resonant frequency there is a decrease in the signal reflected from the antenna which shows up as a spike in the detector output; this output may be displayed on an oscilloscope (see figure 1).

If the sweep generator frequency output is linear (or at least repeatable) in time then the resonant frequencies may be measured by measuring the time interval between the output of a reference cavity (tuned to a frequency  $f_0$ ) and the output of the experimental vessel (see figure 1). For example, if the sweep rate  $r$  is linear, the resonant frequency of the fundamental mode,  $f_1$ , is given by

$$f_1 = f_0 + r (t_1 - t_0) . \quad (2)$$

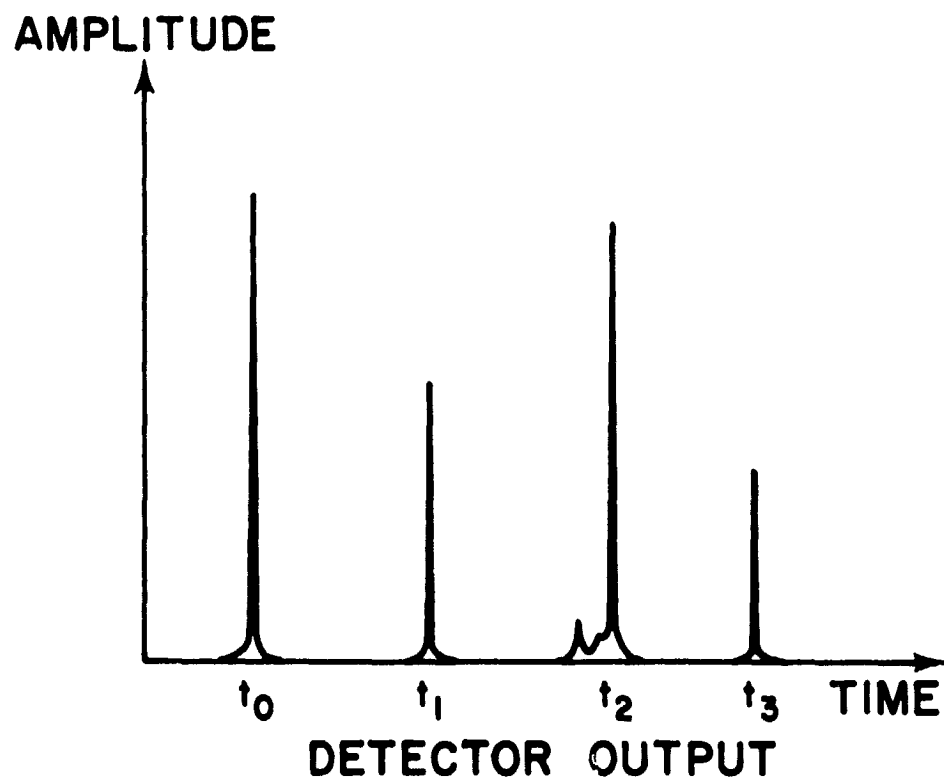
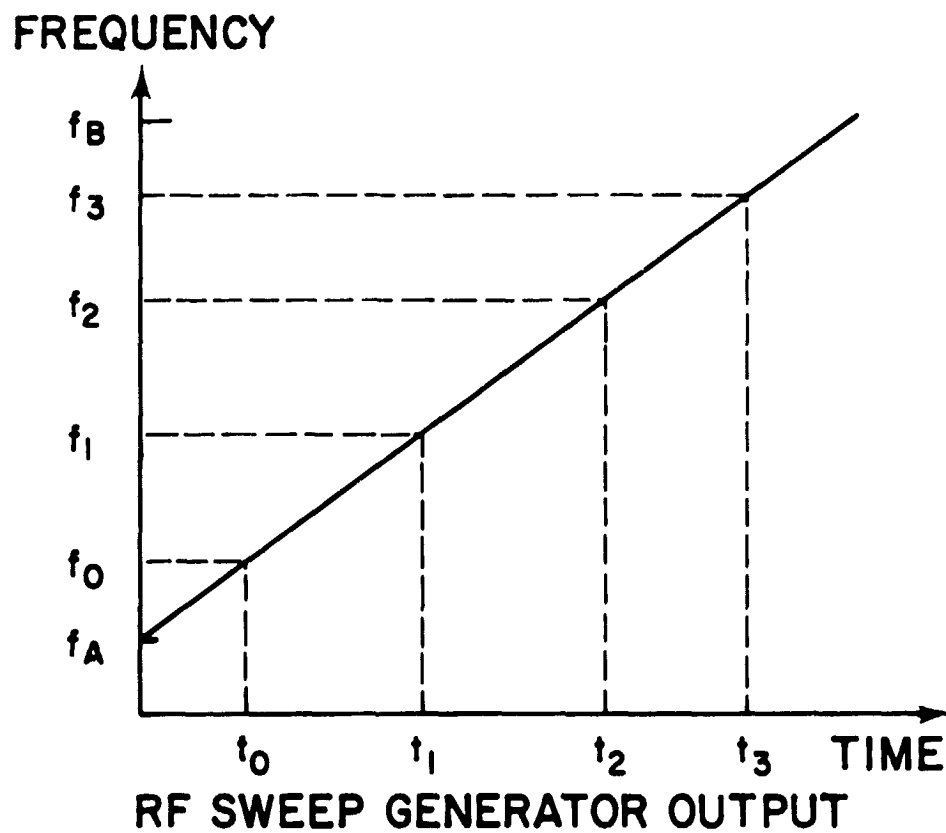


Figure 1. Input and output of RF cavity.

Since  $f_0$  and  $r$  are fixed,  $f_1$  is determined by the time interval  $(t_1 - t_0)$ ; this then is a measurement of the resonant frequency in the "time domain".

The time intervals can be measured using a digital clock and a counter. The clock may be triggered in a start and stop process by signals coming from the reference cavity and experimental vessel, respectively. This is shown schematically in Figure 2. The details of the signal conditioner and data acquisition system are contained in Appendix B.

If the detectors are to start and stop the clock in a precise manner then the pulse must be very sharp and narrow so that the signal in real time comes precisely at the time the RF generator is at the resonant frequency. The narrowness of the pulse is related to the  $Q$  of the cavity which is defined by

$$Q_n = \frac{f_n}{\delta f_n} \quad (3)$$

where  $\delta f_n$  is the width of the spike at the half power points. For example, if  $Q_n$  is 10,000 then there will be about a 0.01 percent uncertainty in the measurement of  $f_n$ .

The  $Q_n$  for several of the resonant modes have been measured in detail for a 19 inch diameter copper sphere, an 18 inch diameter stainless steel sphere and a 5 foot diameter stainless steel sphere. The measured  $Q$  values range between 6,200 and 91,000. The detailed results are contained in Appendix C.

The conclusion of Appendix C is that the  $Q$ 's are high enough to accurately measure the resonant frequencies by the time domain technique even for the large vessel; and it is reasonable to use small vessels for scaling experiments on ultimate large tank configurations.

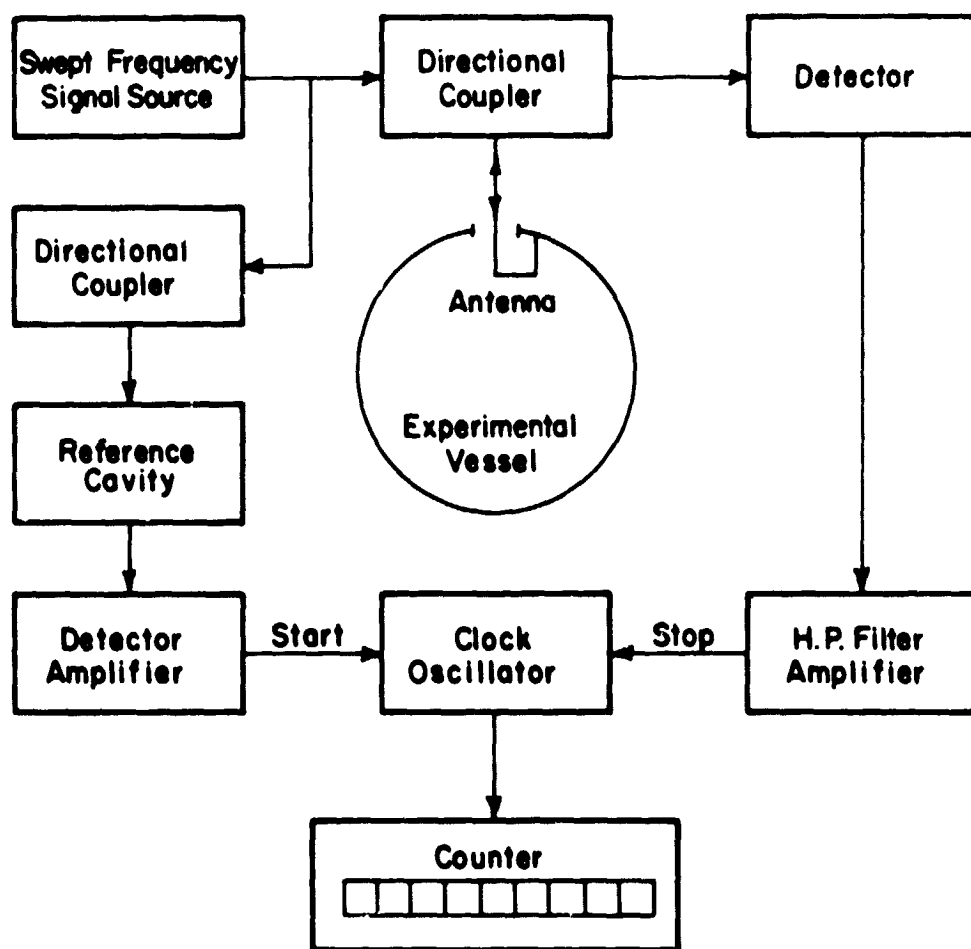


Figure 2. Conversion of resonant frequencies to the "time domain."

## THE RF ANTENNA

One of the attractive features of the RF technique is the simplicity of the internal tank hardware, which is simply a small loop of wire. Figure 3 shows two of the antenna configurations which have been used in the experimental vessel; these are connected to high pressure coaxial feedthroughs.

The straight wire antenna is the "TM probe". It generates only the TM modes. The straight wire is simply an extension of the center conductor of the coaxial feedthrough.

In the loop antenna, the center conductor is bent into a U-shape about  $3/4$  inch by  $3/4$  inch and the end is grounded to the outer conductor. This antenna will generate both TE and TM modes.

The coupling of the antenna to the cavity (and hence the amplitude of the response) is changed only slightly by changes in the size and shape of the antenna. There appears to be a wide variety of antenna size and shapes which are acceptable for this gauging technique.

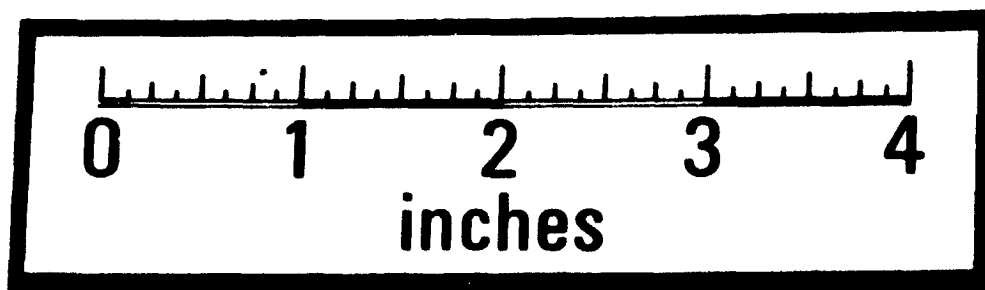
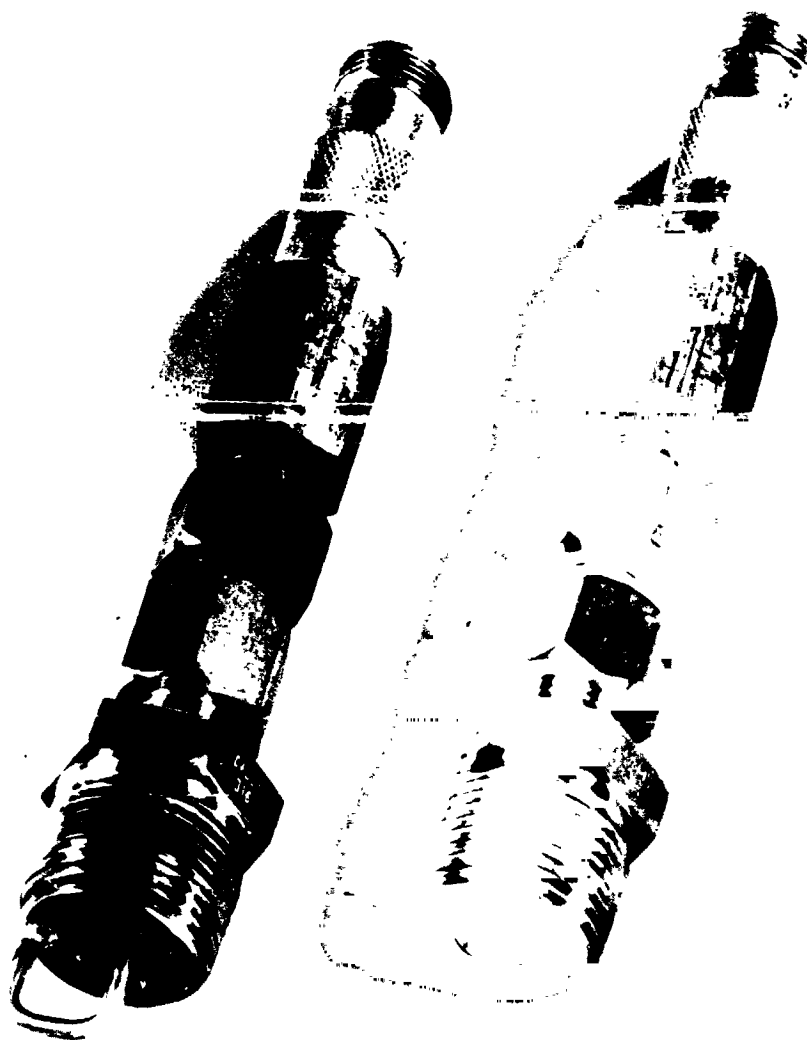


Figure 3. RF Antennas.

## UNIFORM DENSITY FLUID

For a non-polar dielectric fluid of density,  $\rho$ , (a constant throughout the cavity), the resonant frequency of the nth mode  $f_n$  is given by<sup>(1)</sup>

$$f_n = \frac{f_{on}}{\sqrt{\epsilon(\rho)}} \quad (4)$$

where  $f_{on}$  is the empty cavity frequency. Semi-empirically  $\epsilon(\rho)$  can be given implicitly by<sup>(2, 3, 4)</sup>

$$\frac{\epsilon(\rho) - 1}{\epsilon(\rho) + 2} = A\rho + B\rho^2 + C\rho^3 \quad (5)$$

where A, B, and C are constants determined experimentally for each fluid. The  $f_{on}$  are determined experimentally and serve to calibrate the system.  $f_{on}$  depends on the size and shape of the cavity; if the cavity changes size because of thermal contractions or pressure expansions  $f_{on}$  must be adjusted accordingly;  $f_{on}$  may also change if objects are placed in the cavity. To a good approximation, for most situations of interest B and C may be neglected in equation (5); if V is the total volume of the cavity, the mass M in this case may be given by

$$M = \frac{V}{A} \left( \frac{\epsilon(\rho) - 1}{\epsilon(\rho) + 2} \right) = \frac{V}{A} \left( \frac{f_{on}^2 - f_n^2}{f_{on}^2 + 2f_n^2} \right) \quad (6)$$

corrections to this formula for non zero B and C may be applied if necessary. It is seen that possible inaccuracies in the total mass come from four sources:

1. The uncertainties in A, B, and C. This is not a serious problem if the properties data taken by careful capacitance measurements

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1, 2, 3, 4 See references on page 32.

are complete over the ranges of interest. This must be determined for each application involving a specific fluid.

2. The volume of the container. This is usually inferred by weighing a fluid with a known density.

3. The accuracy of  $f_{on}$  which can be measured accurately for an empty cavity; however, if the cavity changes size or shape as a function of fluid density the empty cavity value is no longer valid in equation (6) and corrections must be made to equation 6 to account for this fact.

4. The uncertainties in  $f_n$ . For normal RF frequency ranges and high Q cavities this measurement can be made very accurately. The inaccuracies in converting this frequency into a useful digital or analog signal are presently about 0.2 percent full scale; this can be improved if necessary.

It should be noted that the factors 1, 2, and 3 above may be bypassed by direct calibration of  $f_n$  vs  $M$  using a gravimetric weigh system. However, this is not practical to do for every system. The purpose of this work, then, is to examine the validity of equation (6) (and possibly corrections to equation (6)) for general use in a system where the four accuracy factors may be adequately evaluated. This will be described for the cases of nitrogen and hydrogen. It is anticipated that oxygen will also be evaluated in the near future. The general approach will be to directly calibrate  $f_n$  vs  $M$  using a gravimetric weigh system for measuring  $M$  and a calibrated reference cavity for measuring  $f_n$  and comparing these results with equation (6) using the "time domain" method of measuring  $f_n$ . The amount of any possible density variations will be inferred from an array of thermocouples attached to the cavity.

#### Data Reduction and Readout

Although equation (6) is fairly complicated in form it is surprisingly linear over the density range of interest. A least squares linear fit



to equation (6) for the density range from zero to the normal boiling point shows a maximum deviation of 2.45 percent of the total range for oxygen, 2.2 percent for nitrogen and 1.3 percent for hydrogen. Thus the frequency vs mass relation can be expressed to a good approximation by

$$f_n = f_{on} (\alpha + \beta \rho) \quad (7)$$

where  $\alpha$  and  $\beta$  are constants depending only on the fluid. This simple linear readout may be sufficient for many applications. Quadratic fits of the form

$$f_n = f_{on} (\alpha + \beta \rho + \gamma \rho^2) \quad (8)$$

may be found which give less than 1 percent error for all fluids (see Appendix E).

#### Experimental Data for Nitrogen and Hydrogen

Preliminary experiments were started with liquid nitrogen to check out the experimental apparatus and the theory of uniform density fluids. The pressure was raised to 500-700 psi and the experimental vessel was agitated to achieve nearly uniform density. The frequencies were measured directly using the reference cavity; the line of the reference cavity was shifted to coincide with the resonant line on the face of the oscilloscope. Figures 4 and 5 show the data taken in this manner for the  $TM_{011}$  and  $TM_{021}$  modes respectively. It is seen that the data are nearly linear (slightly concave downward) and corresponds very well to the theoretical form given by equation (6). We believe that the occasional point which deviates from the theoretical curve is due to non-homogeneous density "stratification." We will give some preliminary results on stratification for the case of hydrogen.

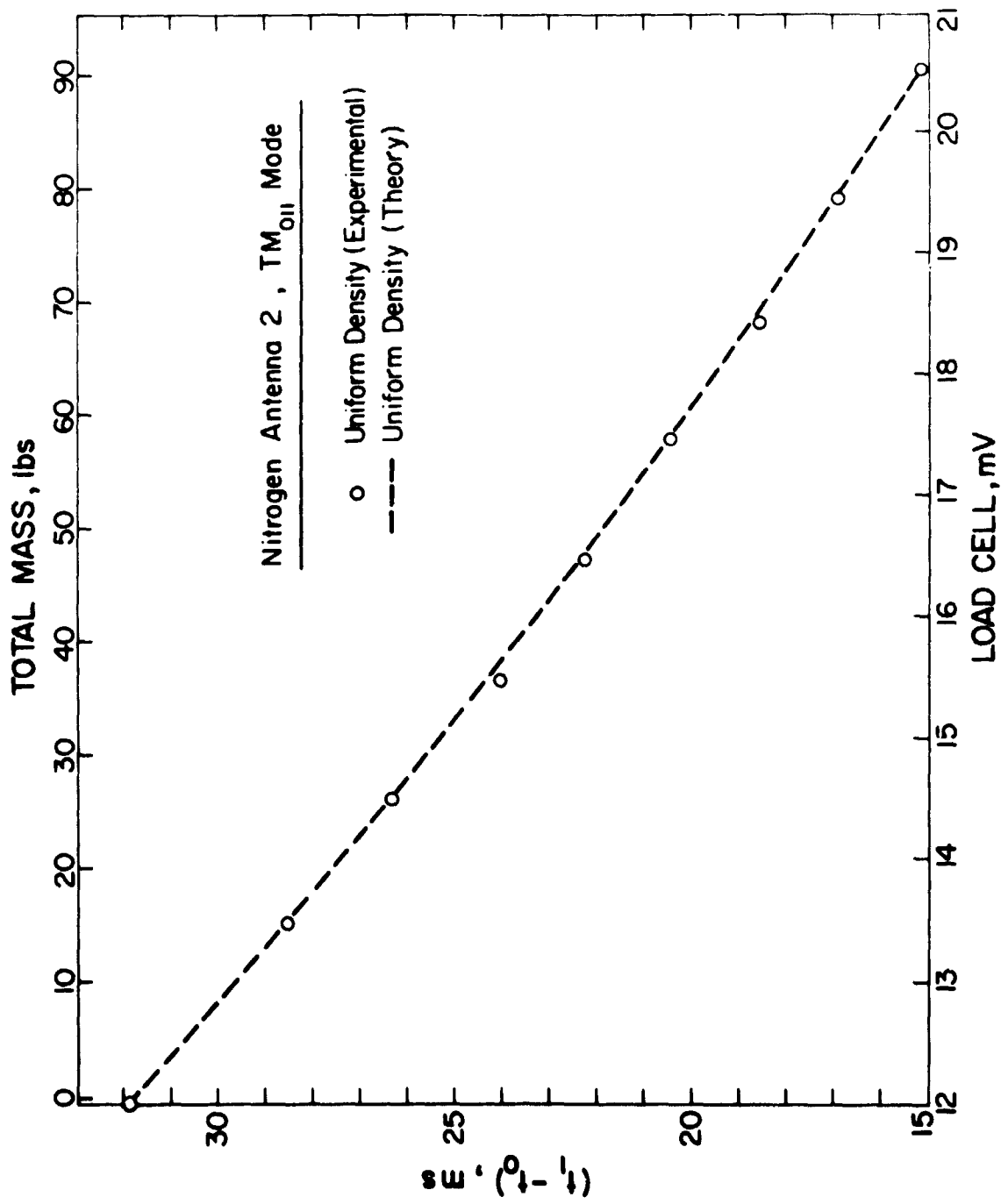


Figure 4. Mass gauging of uniform density LN<sub>2</sub> using the TM<sub>011</sub> mode.

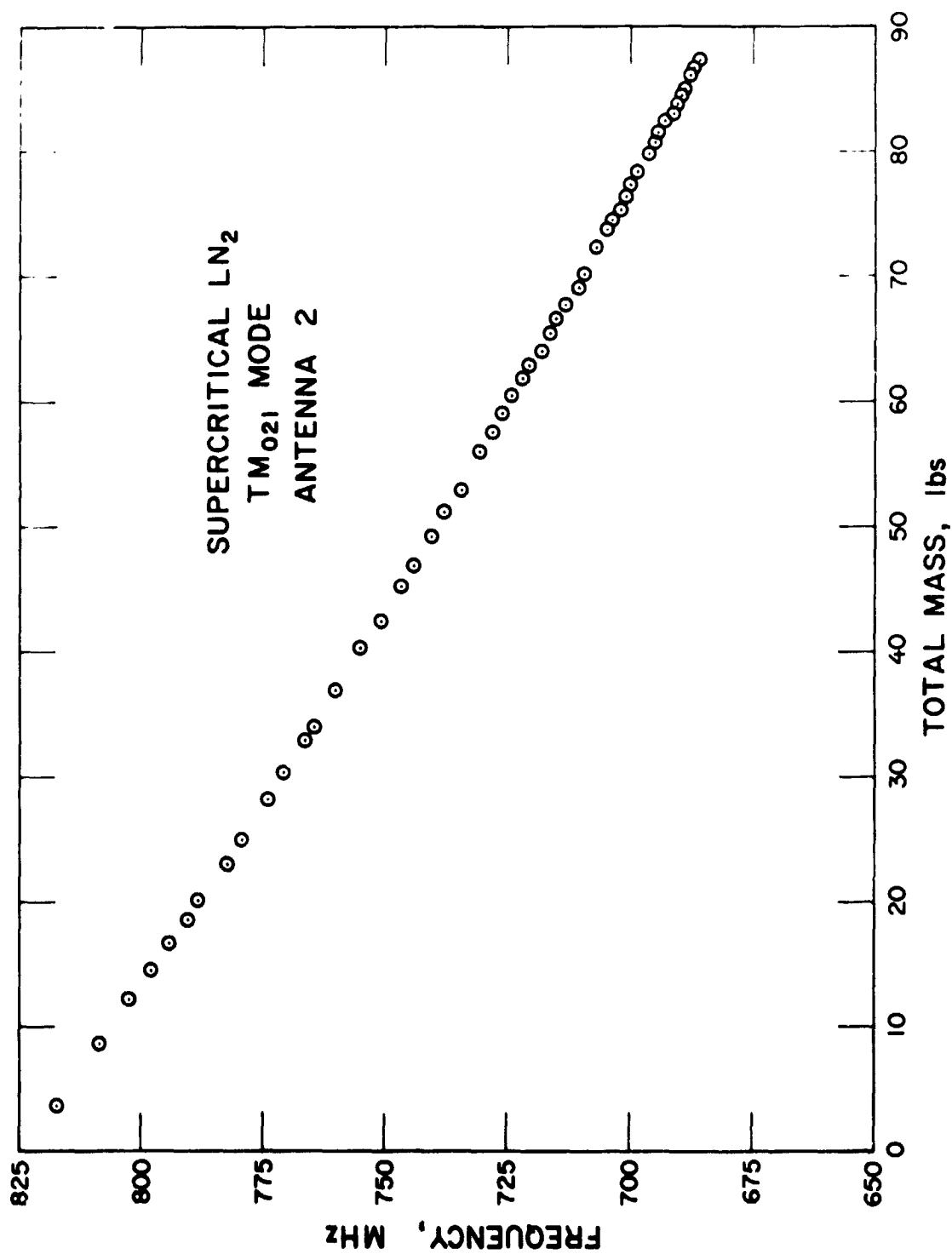


Figure 5. Mass gauging of uniform density LN<sub>2</sub> using the TM<sub>021</sub> mode.

The hydrogen data were taken with the data acquisition system described in Appendix B. The resonant frequencies for the first five fundamental modes were converted to milliseconds in the time domain (as indicated in equation 2) to the nearest 0.01 msec. These times were digitized by an internal clock and stored on magnetic tape. The tape was used to punch data cards which contain a run code, the time conversion of the resonant frequencies, and the scatter in the time domain data. The scatter in the time domain data (except for an occasional noise spike) was  $\pm 0.01$  msec corresponding to the first significant digit. The data cards also contain the pressure, temperature and mass data corresponding to the run identification number. Computer plots of the data for the  $TM_{011}$  and  $TM_{021}$  modes are shown in figures 6 and 7 respectively. A least squares fit of these data to equation (6) gives a  $3\sigma$  (99.9% confidence level) for the  $TM_{011}$  mode of 1.2 percent, the maximum deviation of the data points taken is 1.10 percent. Further details of the data analysis, and also the accuracy statements are contained in Appendices D and E.

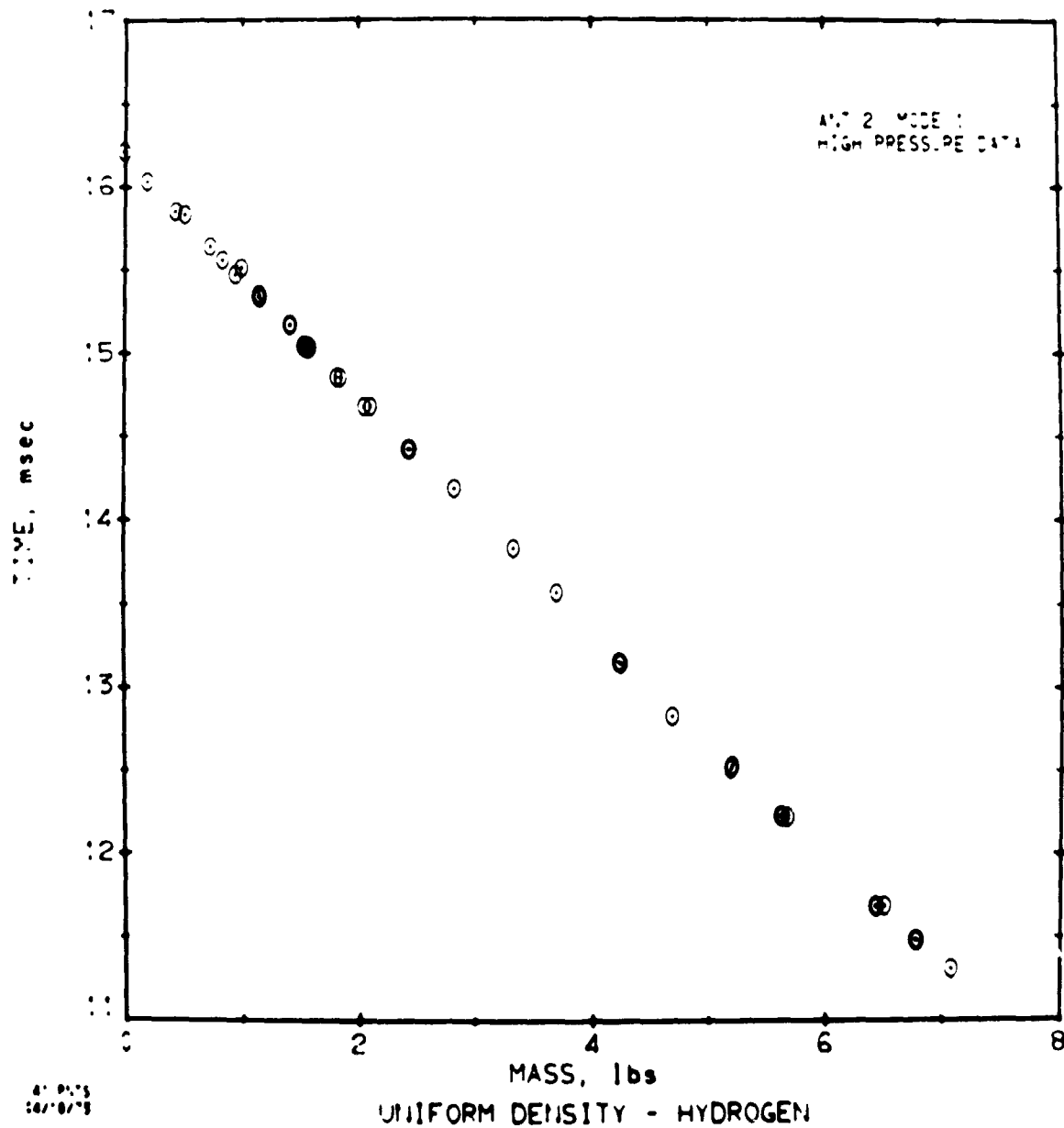


Figure 6. Mass gauging of uniform density  $LH_2$  using the  $TM_{011}$  mode.

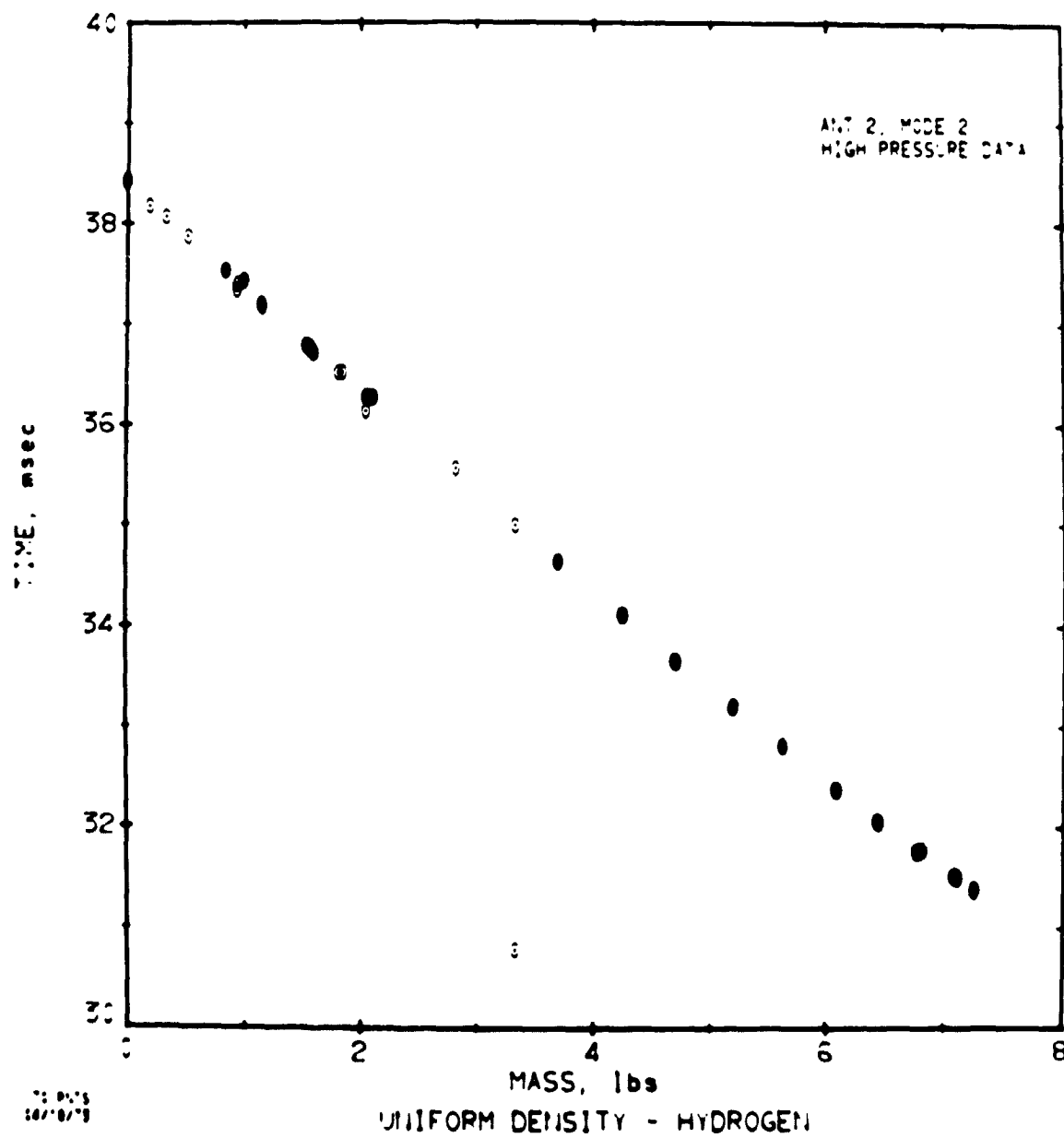


Figure 7. Mass gauging of uniform density  $LH_2$  using the  $TM_{021}$  mode.

## EFFECTS OF STRATIFICATION

In the data described above, every effort was applied to achieve uniform density; however, a few preliminary data have been obtained concerning the effects of stratification on a single resonant frequency. The following cases (in which the top and bottom temperatures were not equal) gives a rough idea of the effect of stratification on the  $TM_{011}$  mode for supercritical hydrogen at 350 psi:

Case I -	Top temperature	= 47.5 K
	Bottom temperature	= 27.5 K
	Top density	$\sim 1 \text{ lb/ft}^3$
	Bottom density	$\sim 4 \text{ lb/ft}^3$
	Average density	$\sim 3.02 \text{ lb/ft}^3$
	Measured weight (load cell)	= 5.35 lbs
	Inferred weight (from fig. 6)	= 5.55 lbs
	% Full scale error	$\approx 4\%$
Case II -	Top temperature	= 37.5 K
	Bottom temperature	= 33 K
	Top density	$\sim 2.25 \text{ lbs/ft}^3$
	Bottom density	$\sim 3.45 \text{ lbs/ft}^3$
	Average density	$\sim 2.00 \text{ lbs/ft}^3$
	Measured weight (load cell)	= 3.55 lbs
	Inferred weight (from fig. 6)	= 3.45 lbs
	% Full scale error	$\approx 1.5\%$
Case III -	Top temperature	= 31.5 K
	Bottom temperature	= 53 K
	Top density	$\sim 3.8 \text{ lbs/ft}^3$
	Bottom density	$\sim 0.8 \text{ lbs/ft}^3$
	Average density	$\sim 1.07 \text{ lbs/ft}^3$

Measured weight	= 1.9 lbs
Inferred weight (from fig. 6)	= 1.9 lbs
% Full scale error	~ 0.0%

In Case I, the stratification is cold fluid on the bottom, warm on top with a good share of the fluid in between at the colder temperature; definitely not a linear thermal gradient.

In Case II, there is a cold fluid on top, a colder fluid on the bottom and warmer fluid in the middle as inferred from the average density.

In Case III, there is a cold fluid on top, a warm fluid on the bottom with most of the fluid in the cavity at the warmer temperature.

From these cases, it is seen that certain small amounts of stratification may be tolerable; this is because the antenna senses the entire cavity and tends to have an integrative effect over all the mass within the cavity. For larger amounts of stratification, one antenna and one mode may not be sufficient to achieve the desired accuracy, and more information from other antennas or modes may be necessary to properly use the RF technique as a gauge. This situation is discussed in the next section on non-uniform density fluids.



## NON-UNIFORM DENSITY FLUIDS - RF MODE ANALYSIS

Non-uniform (inhomogeneous) density (or dielectric constant) may occur either in a single phase fluid under temperature gradients or in a two phase fluid with a liquid-gas interface. The spacial gradients in the dielectric constant have a diffractive effect on the propagation of the electro-magnetic wave and this changes the shape of the standing wave patterns of each resonant mode. These changes will be different for each mode because of the dissimilar field patterns of the modes. The resonant frequency of the  $n$ th mode will be given by

$$f_n = f_{un} + \Delta f_n$$

Where  $f_{un}$  is the resonant frequency expected for the uniform density case assuming that the total mass in the cavity is spread uniformly over the cavity;  $\Delta f_n$  is the change in resonant frequency due to the non-uniform geometry of the fluid and will be different for each mode even to the extent of being positive or negative depending on the mode.

It is natural to ask: what are the extreme limits of  $\Delta f_n$  as the fluid ranges over all possible configurations? This is a difficult and possibly impractical question to answer. The reason is that for a given mass of fluid, there are two fluid geometries which will give the maximum and minimum values for  $\Delta f_n$ ; but these geometries appear to be very complicated and it is unlikely that they will occur in practice. It is more feasible to talk about the practical limits of  $\Delta f_n$  which are determined by experimentation and calculation of  $\Delta f_n$  for likely fluid geometries.

### Theoretical Approach

Theoretical work which was initiated in early stages of Phase I was directed along two lines. The first was to develop approximation techniques for calculating how fluid geometry affects the resonant

frequencies; the second was to investigate the mode geometries to get a qualitative picture of how fluid location may affect the resonant frequencies.

Earlier work had calculated in closed form the resonant frequencies expected for a two phase fluid with a concentric spherical phase boundary ("zero-g" geometry). This work is contained in Appendix F for reference purposes. Further work on two phase geometry effects must be handled by approximation or numerical techniques. Appendix G is a survey of relevant approximation techniques and how they may be applied to the cavity problem. Several of the examples are worked out for the case of the normal two phase fill geometry. One of the conclusions of this work is that the resonant frequencies are most affected when the dense portion of the fluid moves in and out of the high field region. These high field regions are different for different modes. The field profiles for a few of the lowest order modes are plotted in figures 8 through 13. It is seen that the modes partition the cavity into several distinct high field regions; from this, it is expected qualitatively that an average of the resonant frequencies of the lowest order modes may give a mass value that is relatively independent of the location of the dense fluid. The numerical derivation of these graphs as well as numerical solutions of some of the approximation techniques are contained in Appendix H.

#### Experimental Approach

The experimental system described earlier is designed to record on magnetic tape the resonant frequencies of the first five modes for three different antenna locations; the data for each antenna is collected every 0.1 sec. thus making it possible to study dynamic effects where the fluid is in motion. It is anticipated that this system will give useful information in "zero-g" simulation experiments. Some preliminary data

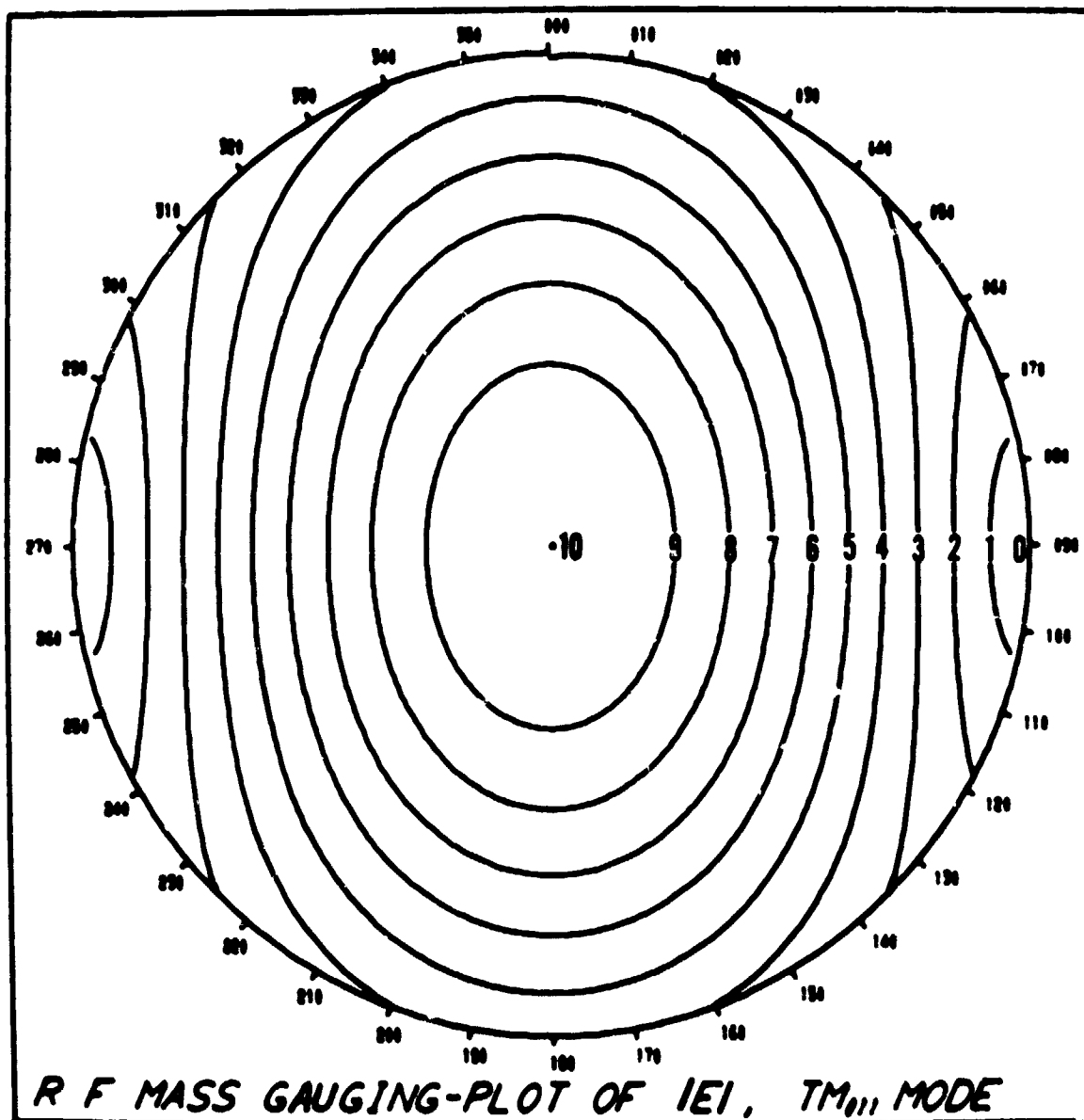


Figure 8. Field magnitude contours for the  $TM_{011}$  mode.

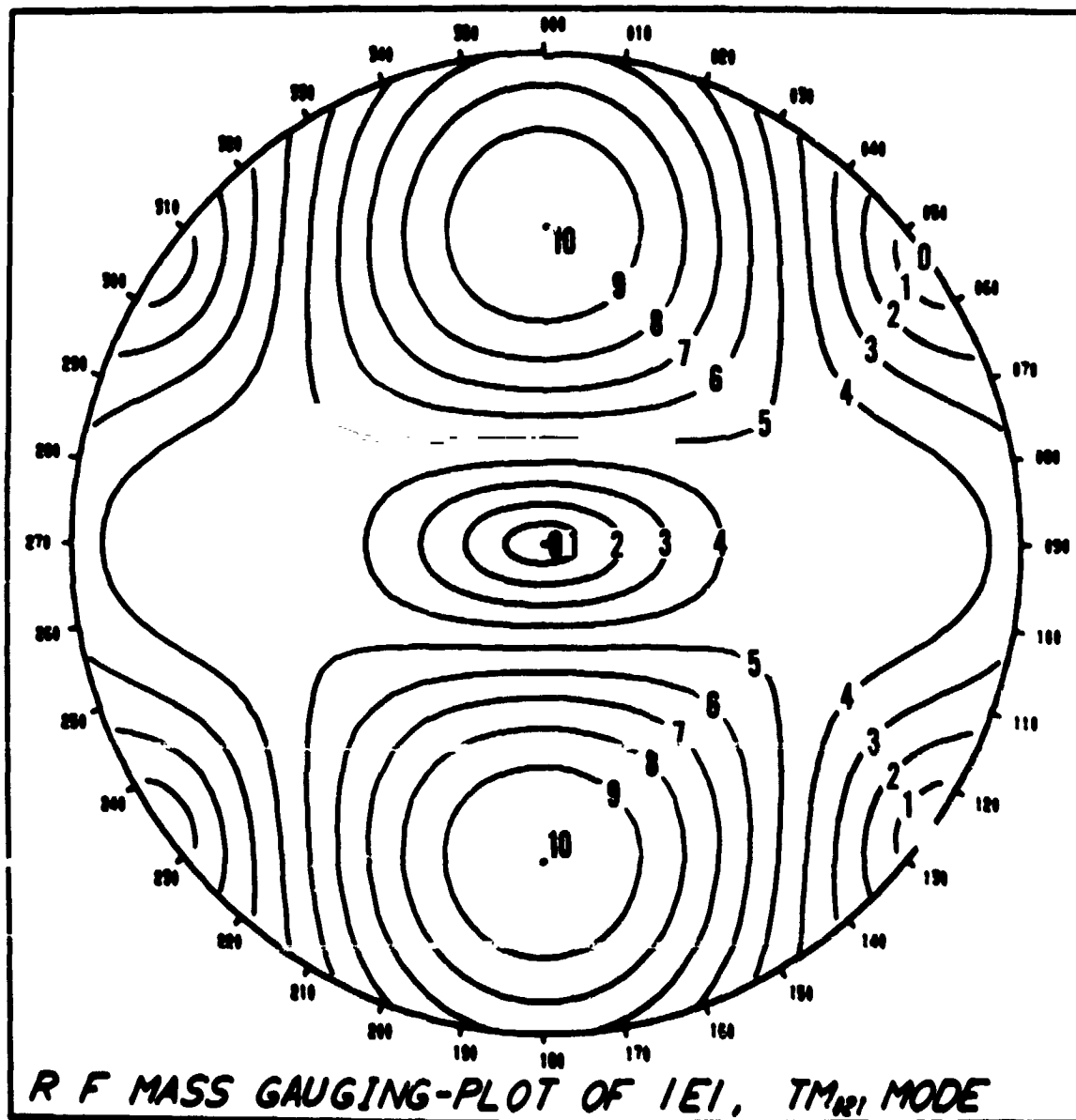


Figure 9. Field magnitude contours for the  $TM_{021}$  mode.

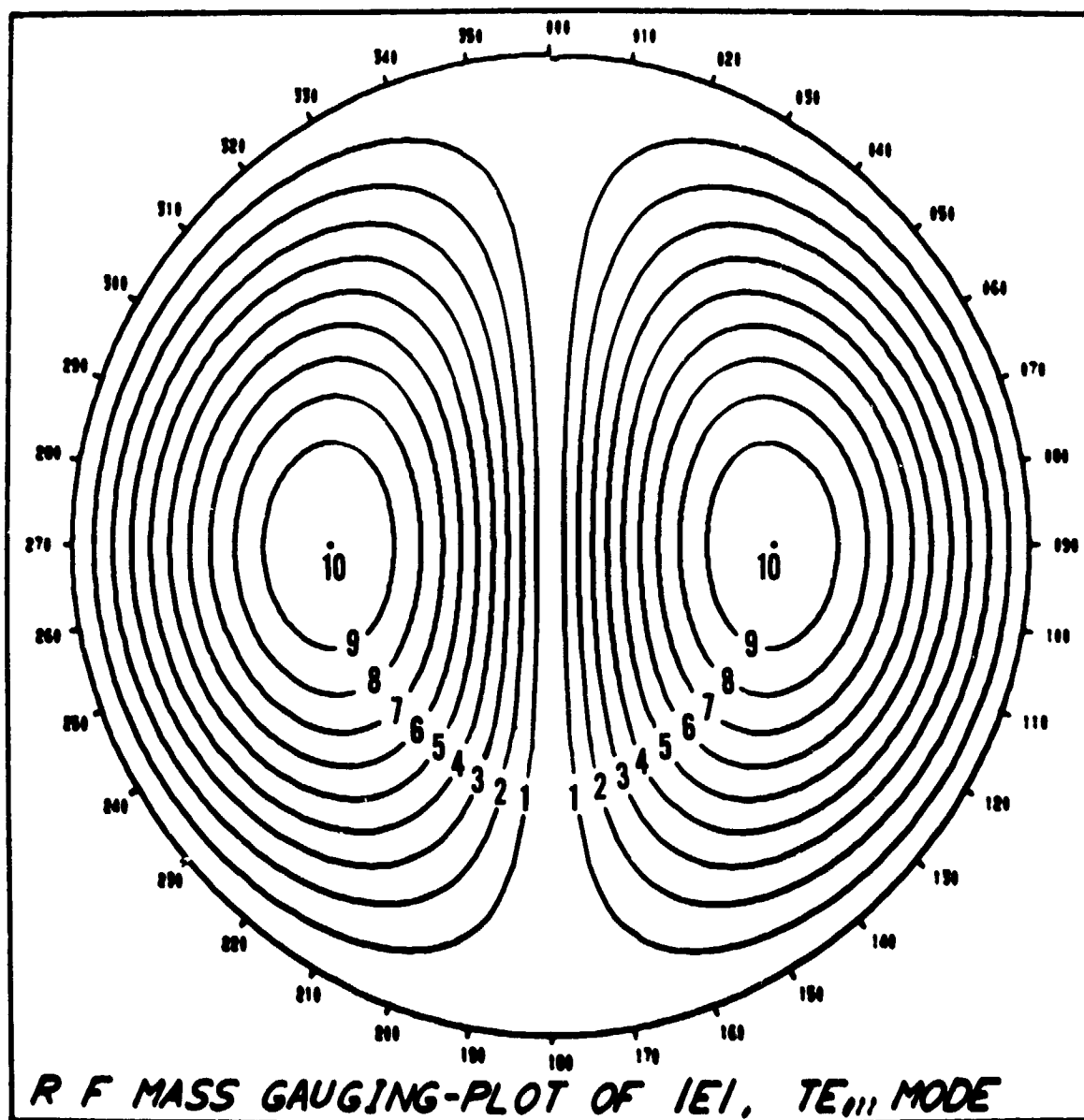


Figure 10. Field magnitude contours for the  $TE_{011}$  mode.

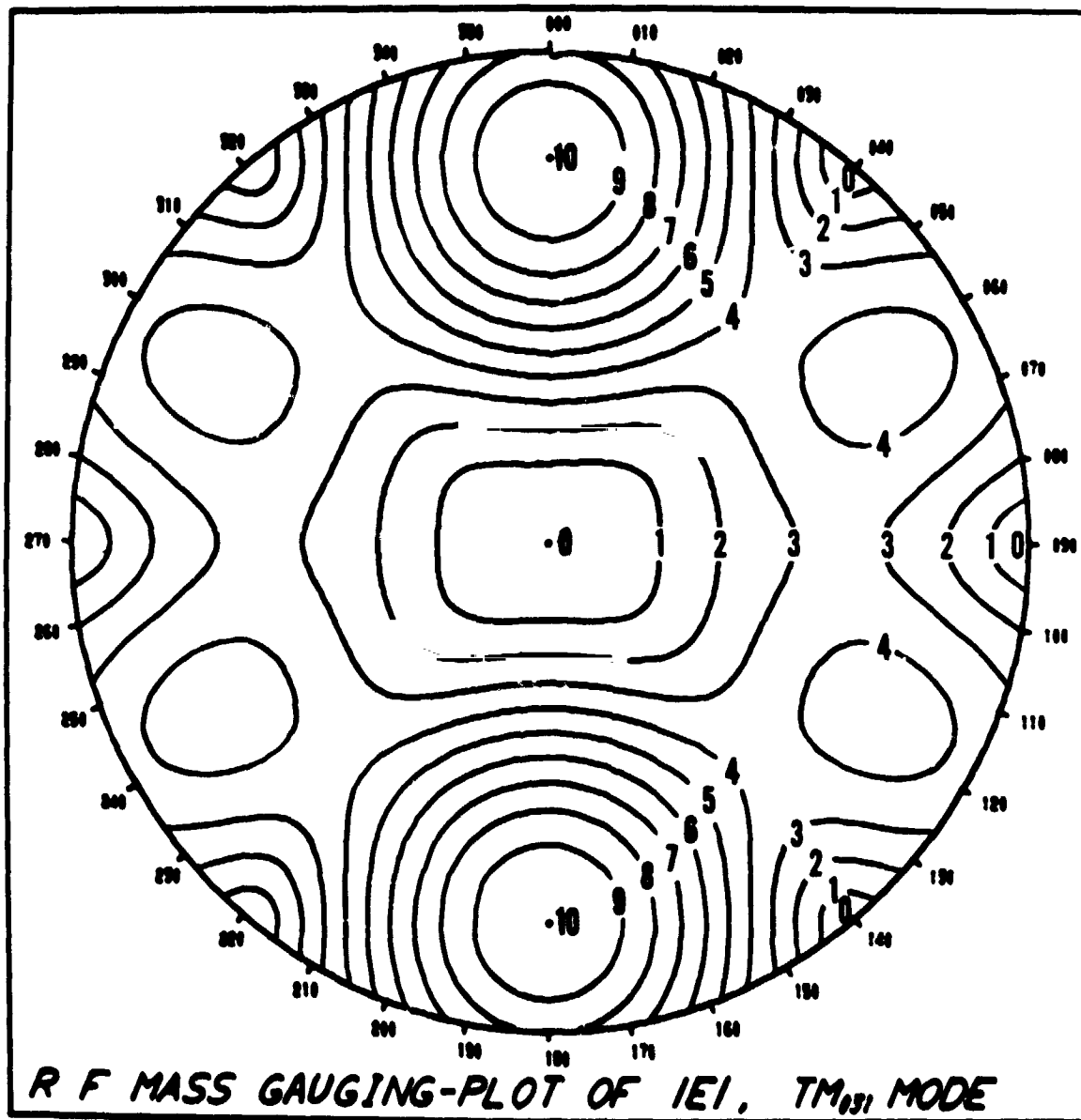


Figure 11. Field magnitude contours for the  $TM_{031}$  mode.

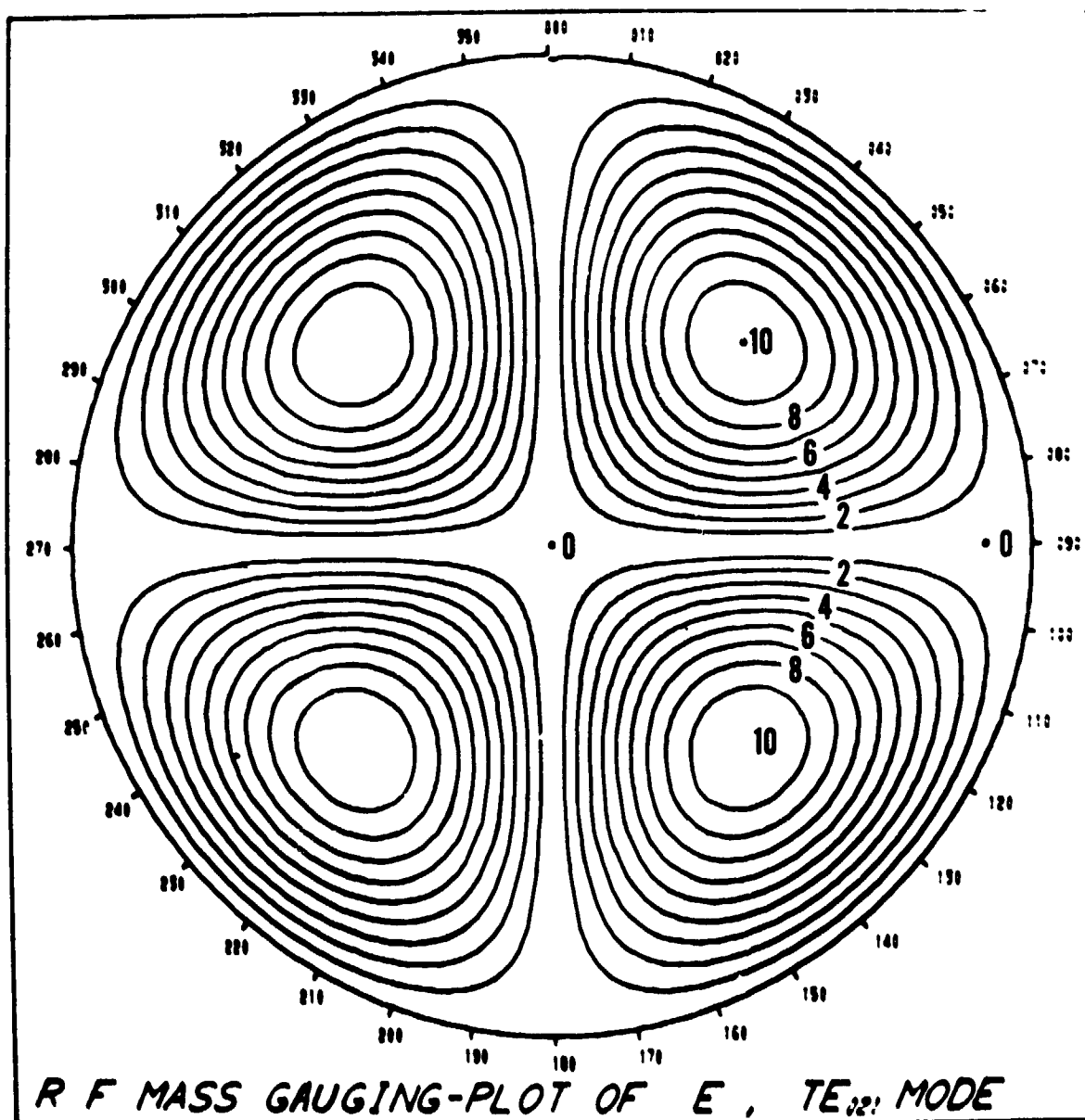


Figure 12. Field magnitude contours for the TE<sub>021</sub> mode.

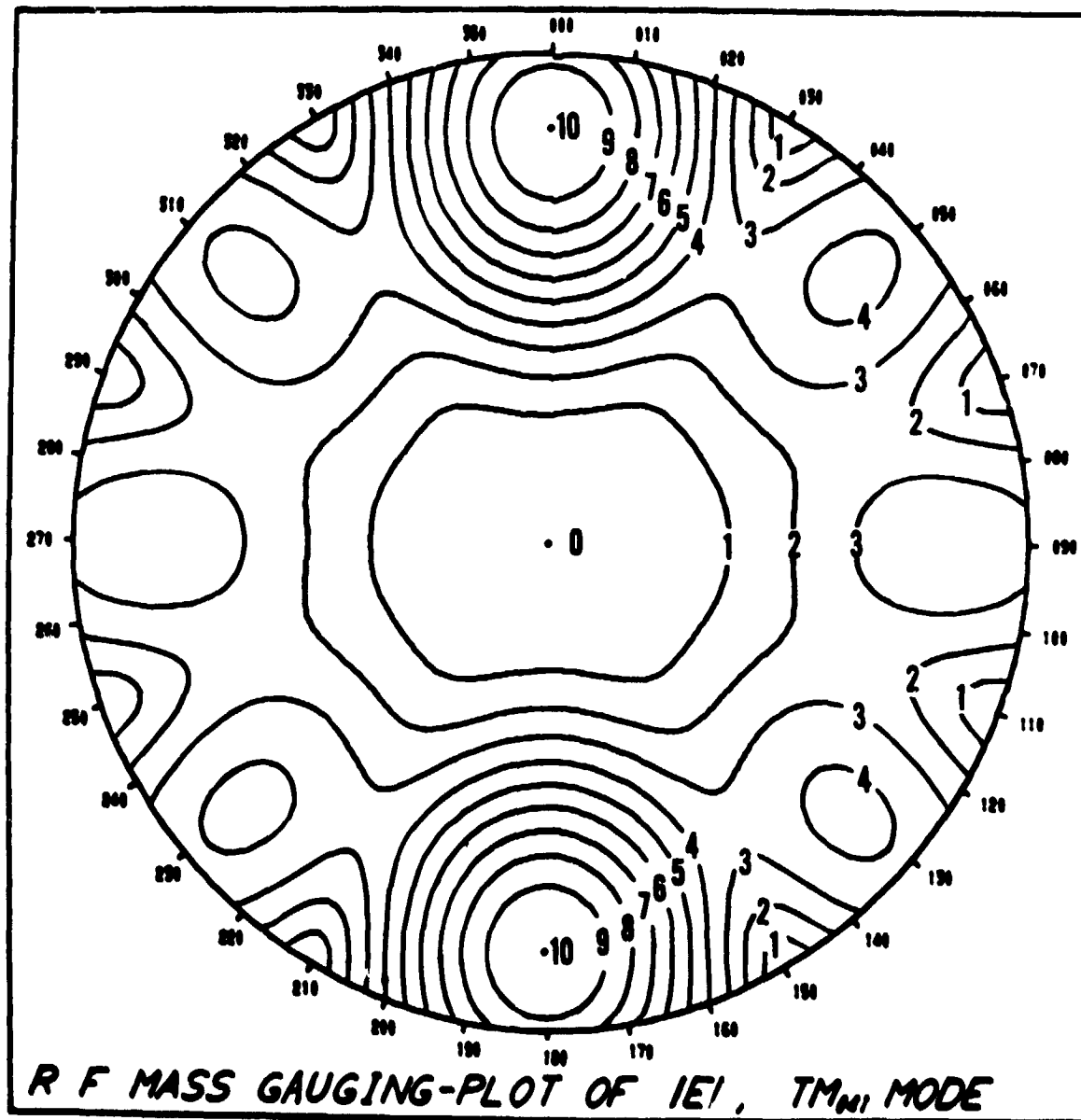


Figure 13. Field magnitude contours for the  $TM_{041}$  mode.



has been taken for normal gravity  $LN_2$  two phase fill. Since the liquid surface breaks the spherical symmetry, it is found that some of the resonant lines split into two or more closely spaced lines; some of the modes do not split. For example, the  $TM_{011}$  mode stays as a single line and the frequency shifts as a function of total mass (see figure 14) giving some idea of the magnitude of  $\Delta f_1$ . Figure 15 shows the response of the  $TM_{021}$  mode which splits into three lines during normal fill. In this case, a straight forward average of the three modes is very close to uniform density curve for this mode as shown in figure 16.

These two cases indicate two alternate methods of gauging the situation of normal fill together with uniform density:

(1) In the case of  $TM_{011}$  mode, the readout can be designed for uniform density and then a correction factor can be applied for the normal fill condition.

(2) In the case of the  $TM_{021}$  mode, the readout can be designed for uniform density, with electronic averaging of the three split resonant lines in the  $TM_{021}$  time frame.

It is reasonable to expect that both of these techniques could be developed to give readout accuracies on the order of 1 percent; however, the averaging technique may be more useful if it can be generalized to tilt geometries and "low gravity" geometries.

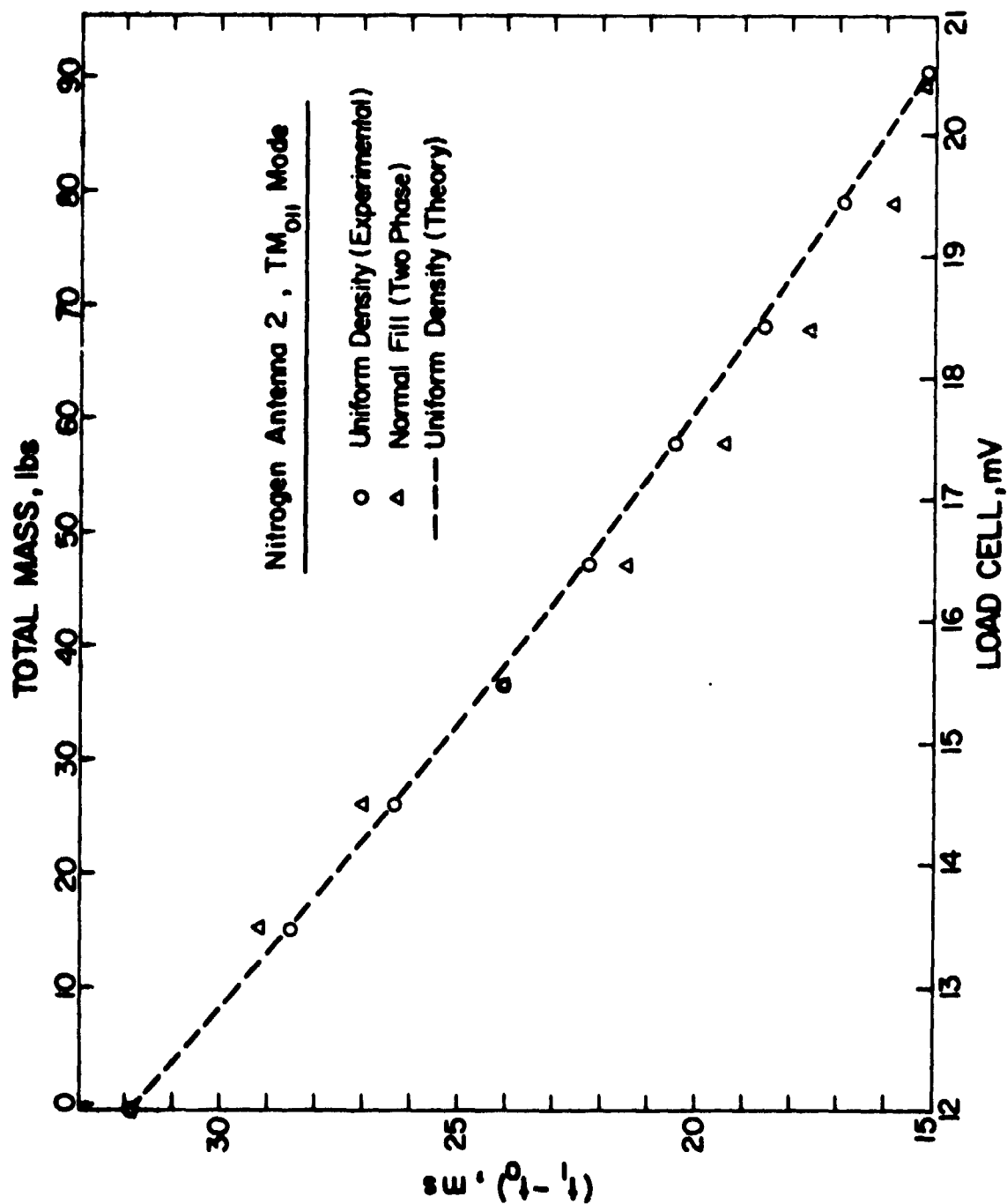


Figure 14. Comparison between supercritical and normal fill for nitrogen, TM<sub>011</sub> mode.

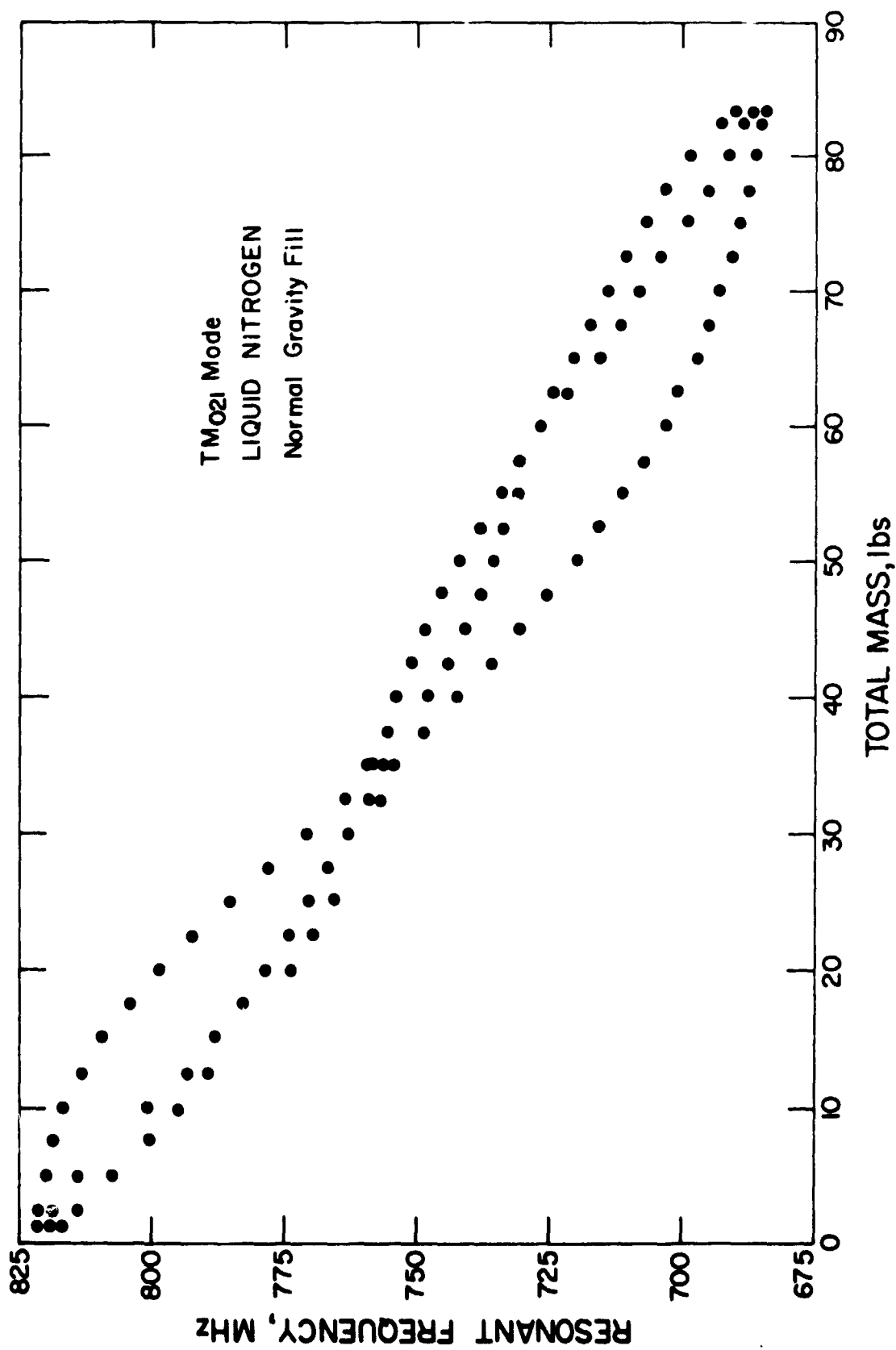


Figure 15. Normal fill data for liquid nitrogen TM021 mode.

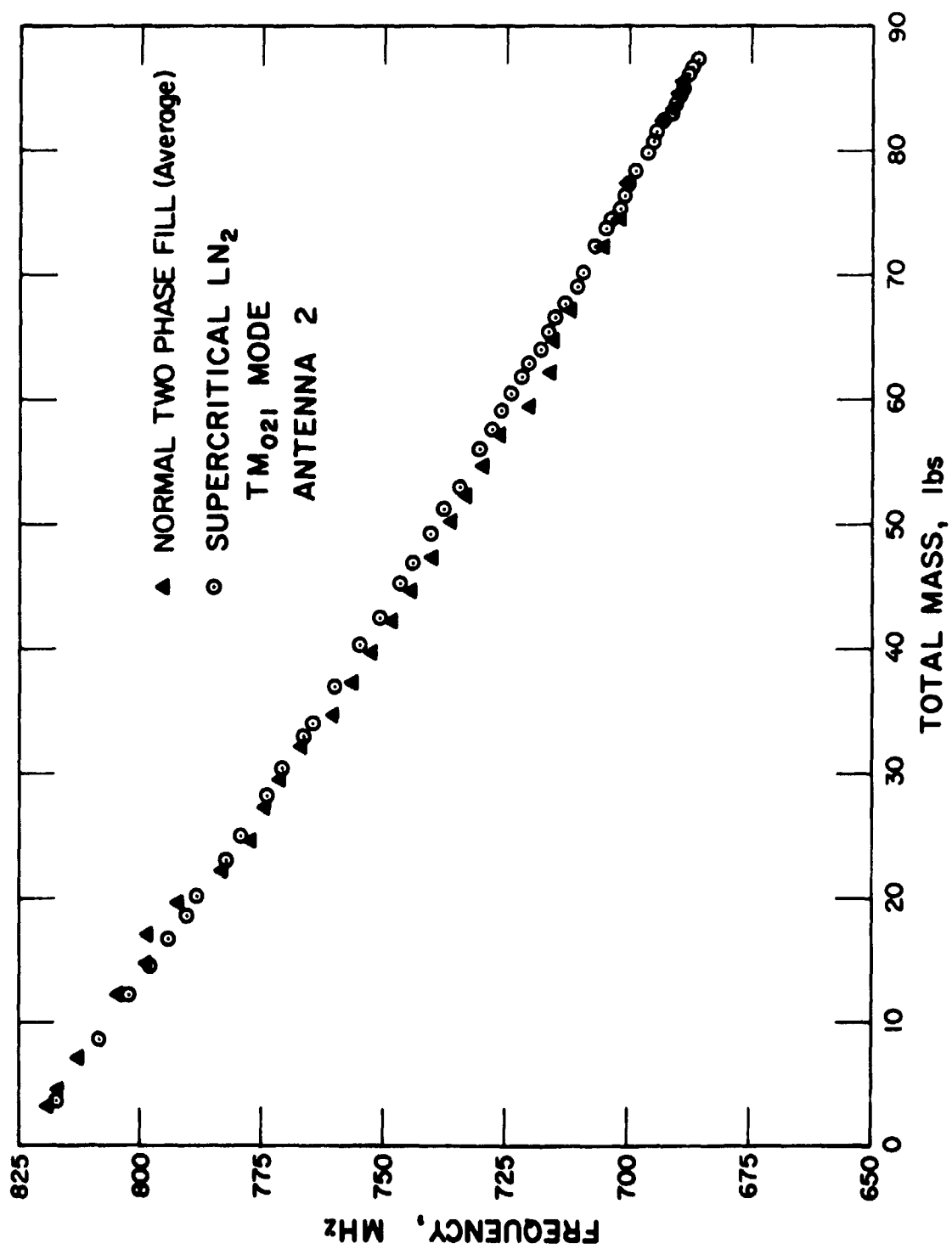


Figure 16. Comparison between supercritical and normal fill (average) for nitrogen, TM<sub>021</sub> mode.

## CONCLUSIONS

We have demonstrated using hydrogen and nitrogen that the RF Mode Analysis Technique is conceptually sound for uniform density fluids and that there are encouraging results for non-uniform density fluids. The results for uniform density fluids should apply to any size or shape of tank as long as the factors listed on pages 8 and 9 can be adequately evaluated. Further work should be done on non-uniform density fluids for the spherical tank and also on tank shapes which are not spherical.

## REFERENCES

1. See Appendix F.
2. Johnston, Andrews and Cole, J. Chem. Phys. 33, 1310 (1960).  
Orantt and Cole, J. Chem. Phys. 46, 697 (1967) (Nitrogen).
3. Younglove, B. A., Journal of Research (NBS), Vol. 76A No. 1  
Jan-Feb 1972. (Oxygen).
4. Stewart, John W., J. Chem. Phys. 40, 3297 (1964) (Hydrogen).

## APPENDIX A

### CRYOGENIC SYSTEMS

The basic system includes a pressure vessel, a heater, a piping system and an insulation system. These components were common to all the test configurations. The normal gravity calibrations utilized a load cell to measure the mass accurately. The design of the flight pallet for the zero gravity tests included an additional storage dewar. The major components of these systems are described.

#### The Pressure Vessel

The vessel is a sphere constructed of 304 stainless steel. It was formed by welding together two spun hemispherical heads. The inside diameter is 17 - 13/16 inches. It was designed and constructed per the A.S.M.E. Code for Unfired Pressure Vessels. The minimum wall thickness is 0.21 inches and the maximum allowable working pressure is 750 psig as defined in this code. The pressure vessel has been hydrostatically tested to 1-1/2 times its working pressure at room temperature. Access ports for instrumentation and piping are provided. All ports, except the port on the bottom, are 3/4" female pipe threads. The bottom port is 1-1/2" female pipe thread to accommodate the heater. All ports are reinforced with welding spuds.

#### The Heaters

Two heater configurations have been used. The first was designed for use with liquid nitrogen and the second for liquid hydrogen. Both were designed to pressurize the fluids to their critical pressures in forty minutes. Both were constructed with temperature independent resistance wire. Pressurization lines as a function of heater power and current are presented in figure A1.

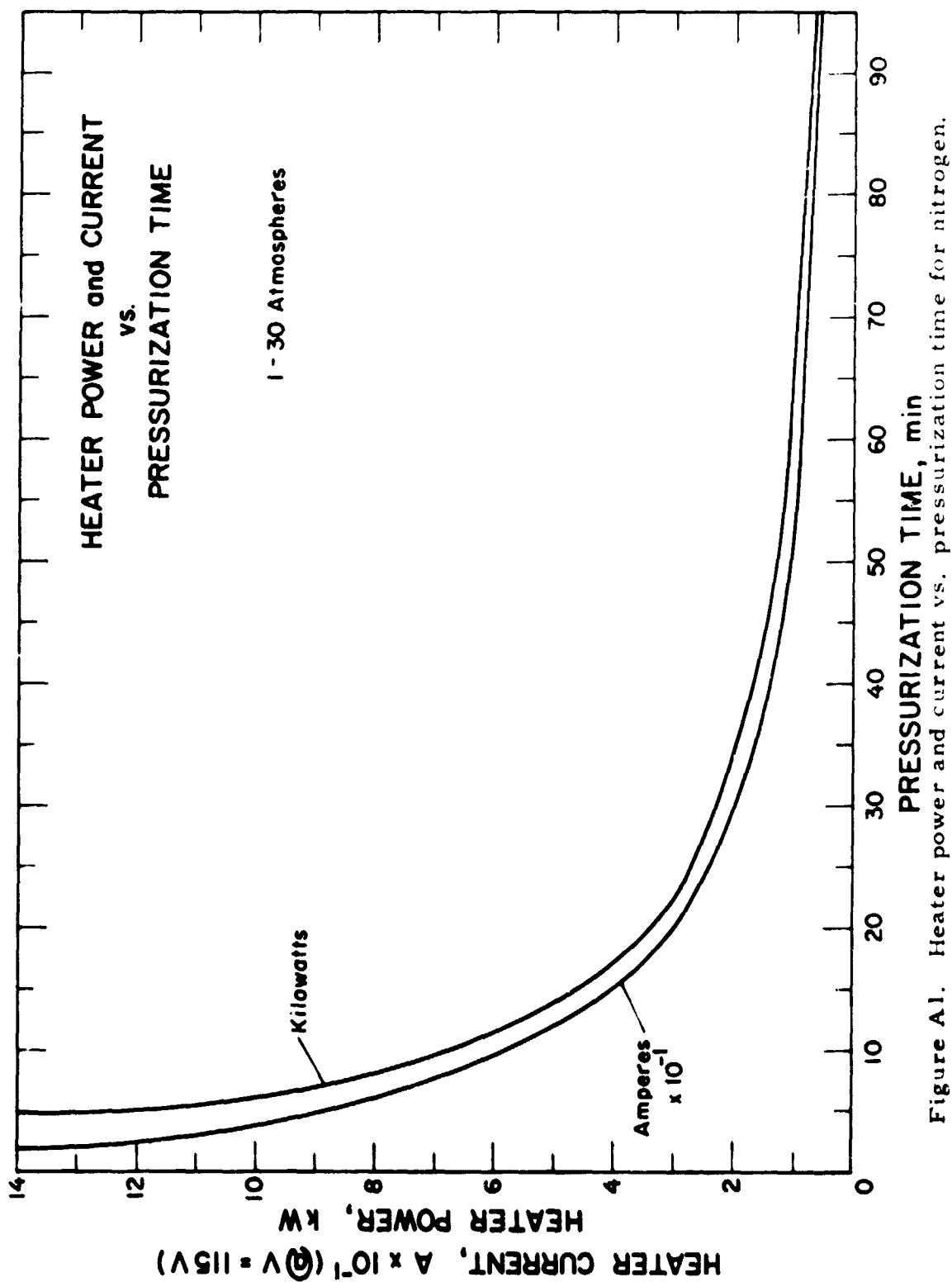


Figure A1. Heater power and current vs. pressurization time for nitrogen.



The nitrogen heater system consisted of two separate one kilowatt heaters that could be separately controlled. Each heater was wound with ten feet of 22 gage wire. Each had a room temperature resistance of 12 ohms. The wire was wound on two separate coils. The coils were then inserted in the bottom access port with the wire in direct contact with the nitrogen.

The hydrogen heater was constructed differently because of safety considerations with the flammable fluid. This heater consisted of one 300 watt coil of wire. It was wound with four feet of 28 gage wire with a room temperature resistance of 20 ohms. The wire was covered with ceramic beads and inserted into a coil of 1/4" O.D. copper tubing. The tubing was then soldered on the outside of a brass spindle which was screwed into the bottom port. The heater element does not contact the hydrogen but heats the pressure vessel from the outside. The copper tubing is purged with helium gas and is electrically grounded.

#### The Normal Gravity Calibration Tests

For these tests, the pressure vessel was hung in a cubical frame constructed of aluminum angle. The piping system and the insulation system was also supported from the frame. The frame, and the systems it supported, was suspended from the load cell. All interface piping and instrument lines were flexible to minimize their effect on the load cell. Figure A-2 is a photograph of this system without the insulation.

The piping was designed for use with liquid hydrogen and was used with liquid nitrogen as well. The system is equipped with a 750 psig

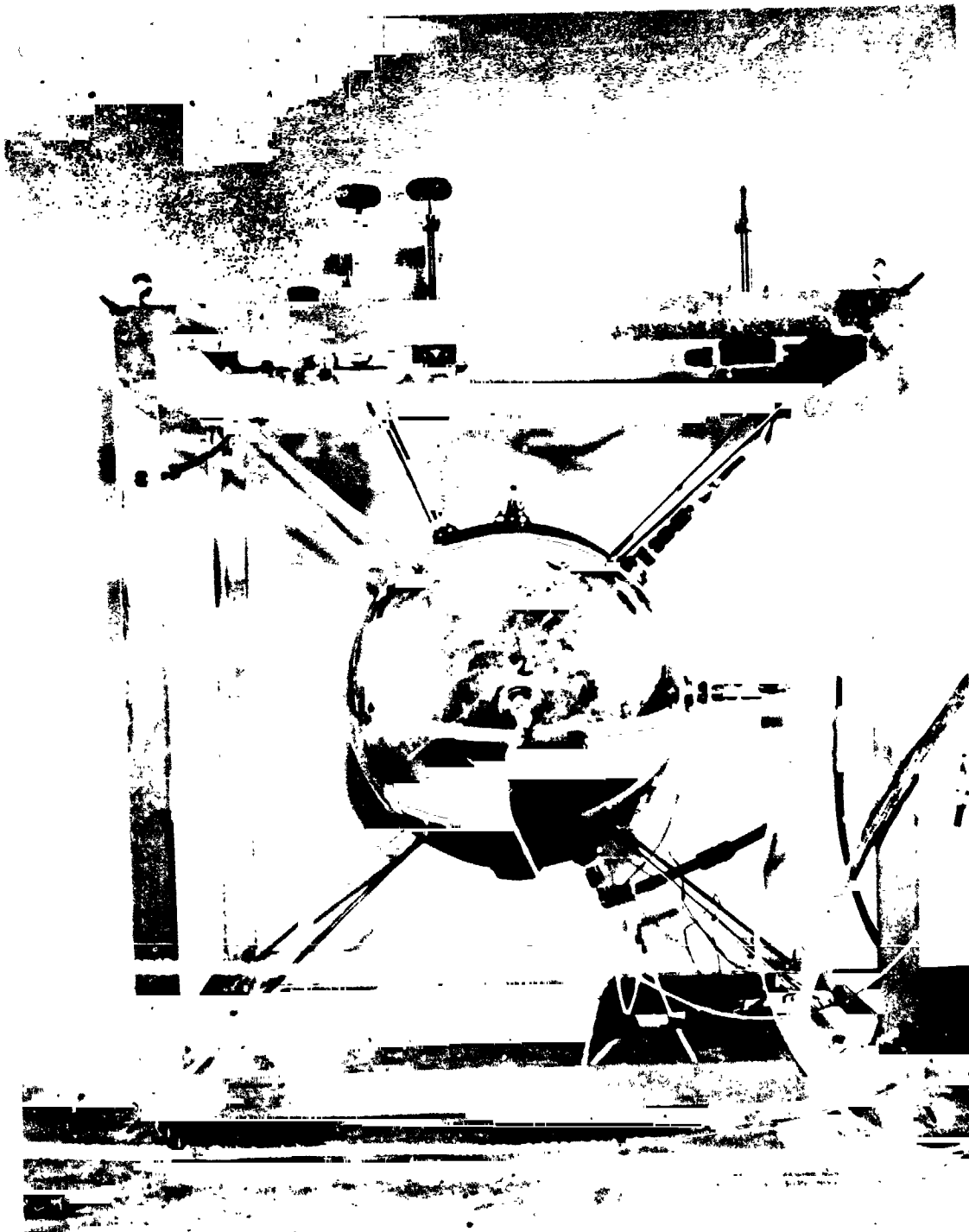


Figure A2. Experimental vessel.

relief valve. The system has been pressure tested to 700 psig with liquid nitrogen as the test fluid and it has been tested to 300 psig with liquid hydrogen.

The insulation system was made up of six inches of fiberglass batting. To prevent cryopumping the fiberglass was covered and sealed with aluminized mylar. The insulation was then purged with helium gas and a slight overpressure was maintained during the tests.

The load cell used was a strain gage type rated at 300 pounds maximum load; it has a load sensitivity of 0.1 mv/lb with a 10 volt excitation voltage. Using a high precision digital voltmeter on the output, the resolution is approximately 0.01 lbs.

The pressure vessel was instrumented with several temperature measuring transducers to determine the temperature and density gradients during the supercritical tests. Nine copper-constantan thermocouples were attached to the outside wall of the pressure vessel at six inch vertical increments. Platinum resistance thermometers were inserted into the fluid at the top and bottom to determine actual fluid temperature.

#### The Zero Gravity Tests

The design of this system utilizes the same pressure vessel, heater and support frame as the normal gravity system. The piping and insulation systems have been changed.

The piping system flow diagram is shown in figure A3. All the valves are manually operated. The design utilizes a common overboard vent and liquid dump, which will be compatible with the aircraft. The system can be refilled and pressurized in flight if necessary.

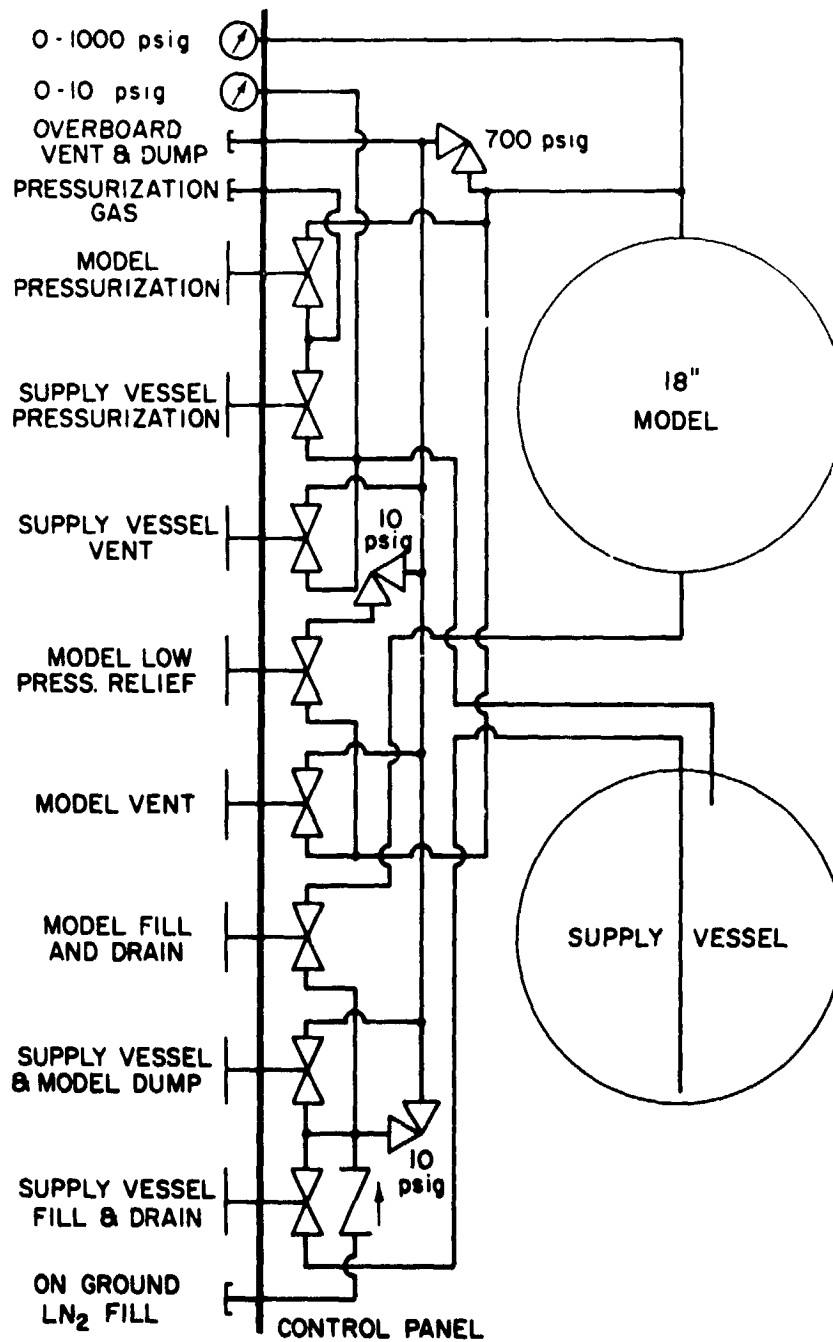


Figure A3. Cryogenic flow system for zero-g simulation tests.

The complete system will be packaged in an aluminum paneled container six feet long by three feet high and three feet wide. The total weight will be approximately 700 pounds.

The system is designed to withstand an acceleration load of 16 g's in any direction. This will be accomplished by using a polyurethane foam insulation system. After the entire system has been assembled in the paneled container, the foam will be poured, filling all remaining space. The foam, then, will not only insulate but will support the system. The outside paneling shall be 3/8 inch aluminum plate.

With this insulation system, the total heat leak will be approximately 15 BTU per hour. The rate of pressure rise, as a result of this heat leak, will be 1.5 psi per hour.

APPENDIX B  
MASS TANK GAGING SIGNAL CONDITIONER AND DATA  
ACQUISITION SYSTEM

Purpose

To condition the resonant frequencies of the mass tank, to measure these frequencies as a function of time and store the data in a magnetic tape unit.

Operation

A radio frequency generator is swept through a designated frequency spectrum of a starting frequency  $f_A$ , to an ending frequency  $f_B$ .

This generator is connected to a cryogenic mass tank. The mass tank has resonant frequencies (modes) which are enhanced each time the tank is energized with the generator sweep of selected frequencies.

The design objective of the data acquisition system and signal conditioner is to measure these enhanced resonant modes with respect to a known frequency and record them.

In the system this measurement is accomplished by using a time measurement technique. The generator sweep is used as a time base for all of these measurements.

A reference cavity tuned to a frequency,  $f_0$ , which is lower than the lowest cryogenic cavity mode, is the reference starting time for the time measurement of the mass tank system.

When the generator energizes the reference cavity tuned to  $f_0$ , a pulse is generated by the cavity. This pulse is used to start a counting sequence using a 100 kHz clock as the basic counting device.

The first resonant mode,  $f_1$ , from the mass tank strobes the counter into a buffer register which loads the data to a shift register.

The shift register then sends the data to a magnetic tape unit. All these data transfers occur before the next resonant mode of the mass tank is generated.

The counter continues counting with its data being strobed into the shift register every time a resonant mode is generated.

Upon completion of the generator sweep, the counter is reset, another antenna in a different location on the mass tank is multiplexed, and the generator initiates a new sweep. Beginning of the sweep initiates a new set of data.

Using this technique, a measurement of mass in the cryogenic tank is obtained as a function of time with respect to the reference cavity tuned to  $f_0$ .

## Data Acquisition Schematic I

### Functional Description

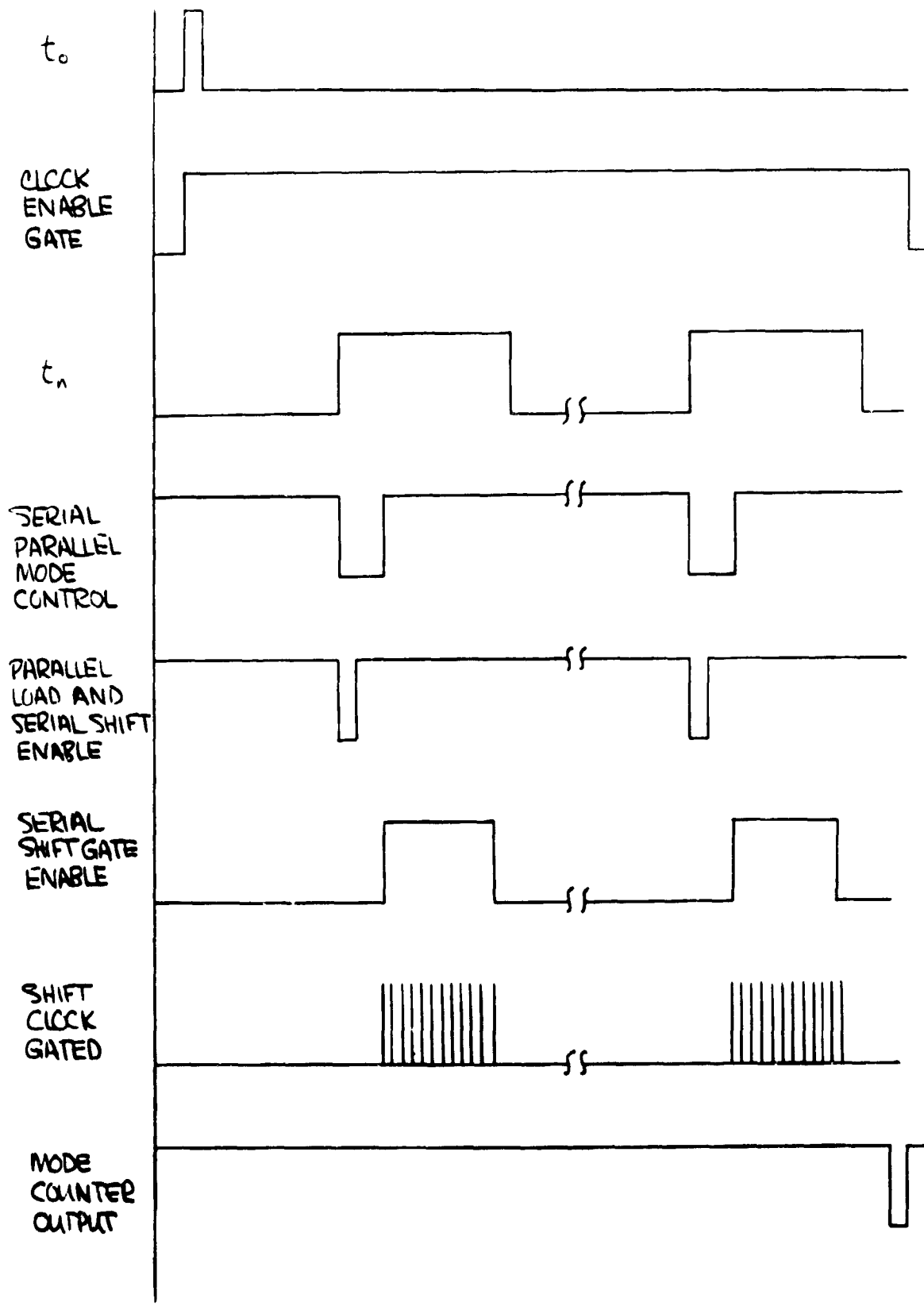
This schematic shows the timing sequences necessary to start the counting sequence and data transfer functions. All timing events are referred to the initiating pulse,  $t_o$ , generated by the reference cavity.

The pulse  $t_o$ , opens the gate to the counter and enables the real time clock to send 100 KHz pulses to the six stage counter.

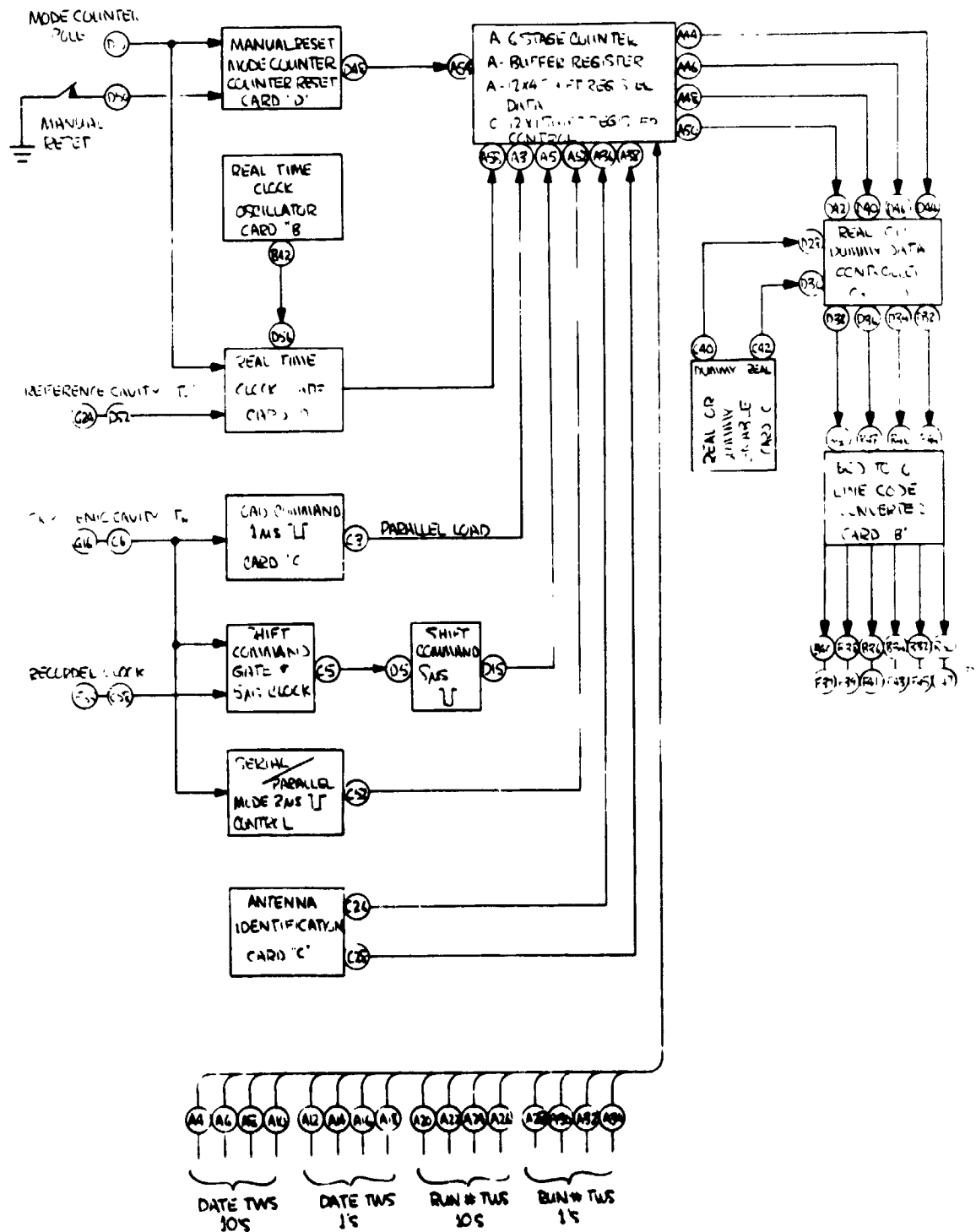
The next timing event occurs when the cryogenic mode pulse  $t_n$  is generated. This pulse generates a strobe command to the storage buffer, a parallel load command to the shift register, a serial-parallel mode control pulse to the shift register, and a shift command gate enable pulse to the serial shift clock.

The mode counter generates a clock disable signal at the end of the last mode to be measured and resets itself.





# DATA ACQUISITION SCHEMATIC I



## Data Acquisition Schematic II

### Functional Description

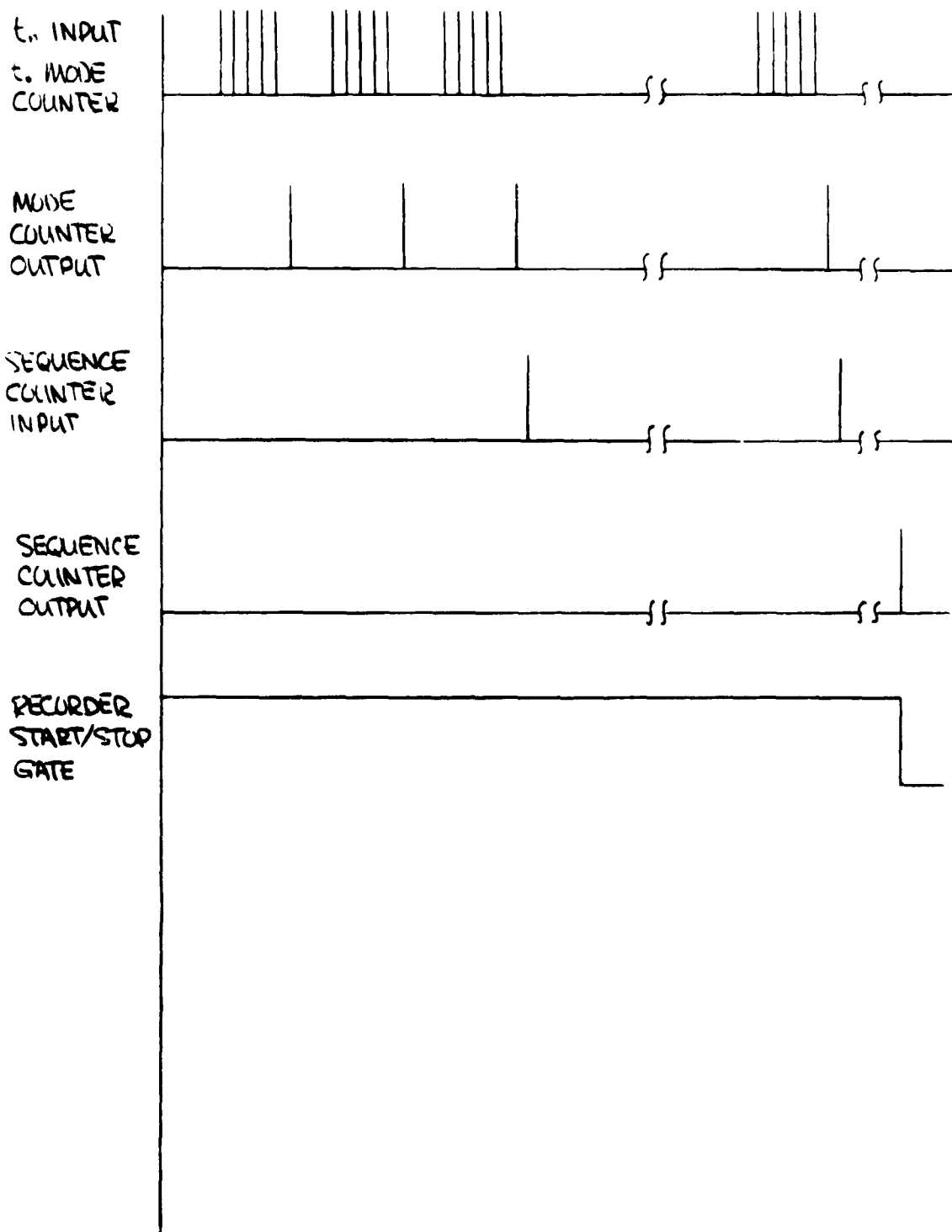
This schematic shows the timing sequence necessary for control of the mode counter, antenna multiplexer, sequence counter and magnetic tape recording unit. All timing events are referred to the mass tank pulse  $t_n$  generated by the modes  $f_n$ .

A recording sequence is initiated by depressing the manual reset and manual start push buttons. The manual reset button resets all counters and controls flip flops to initial status for recording and controlling. The start push button starts the magnetic tape unit.

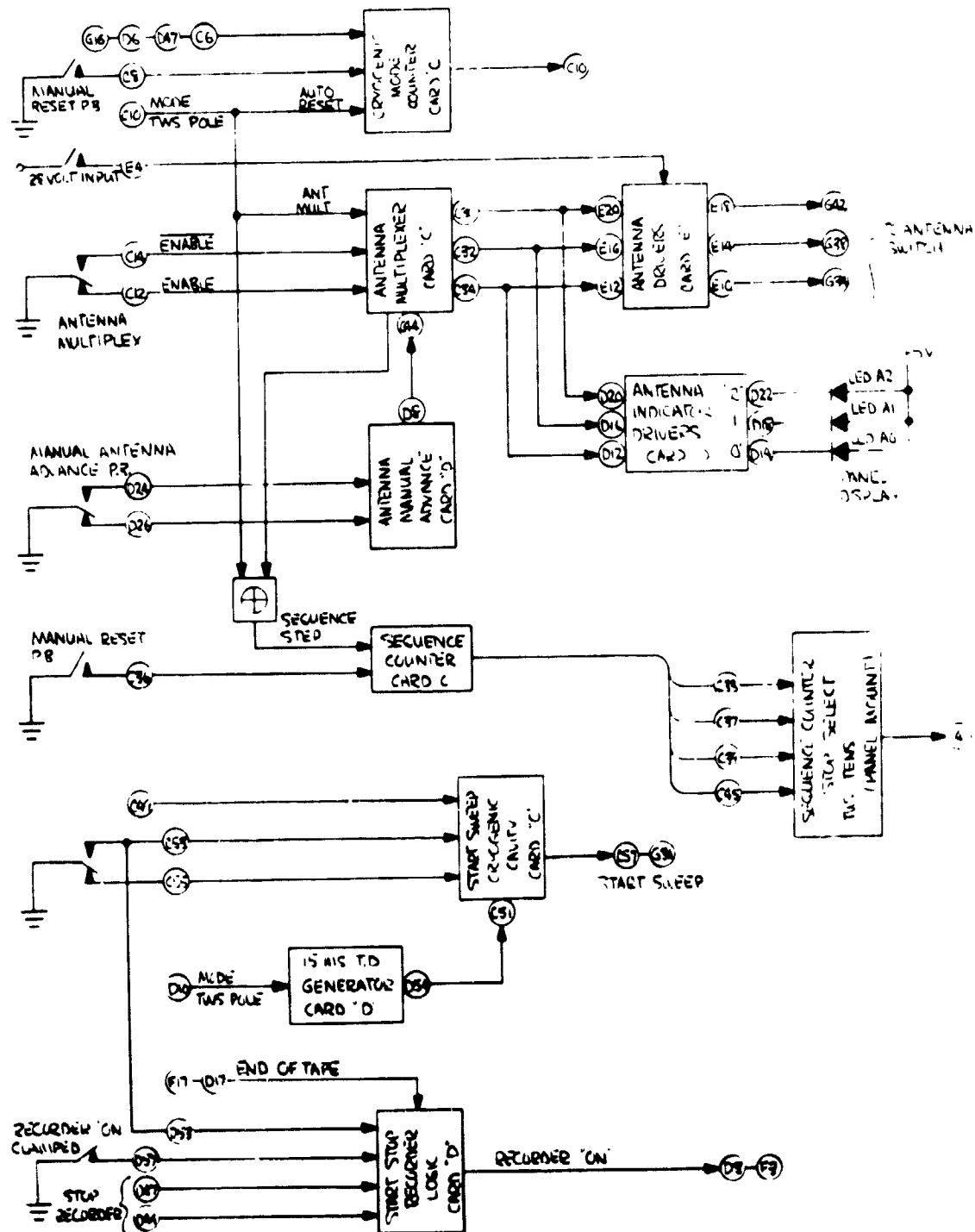
The  $t_n$  pulses are counted by the mode counter until a preselected output is generated in the counter. This output multiplexes the antenna or sends a counting pulse to the sequence counter if the antenna multiplexer is disabled.

The antenna multiplexer generates an output for every complete cycle of antenna multiplexing. This output sends a counting pulse to the sequence counter when multiplexing is enabled.

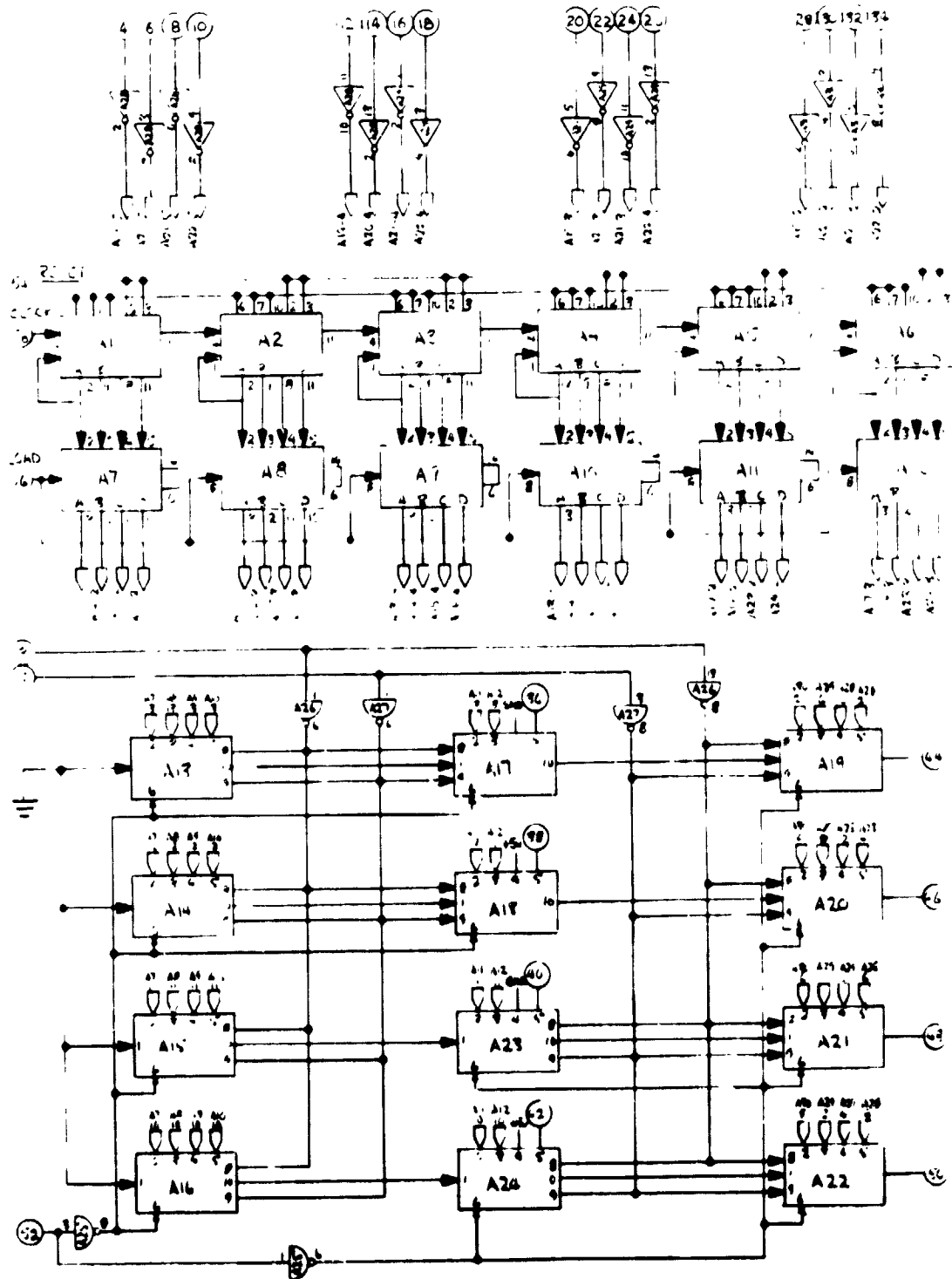
The sequence counter has a preselected number of counts it can accept before it overflows and stops the magnetic tape recorder.



# DATA ACQUISITION SCHEMATIC II



# COUNTER-SHIFT REGISTER CARD

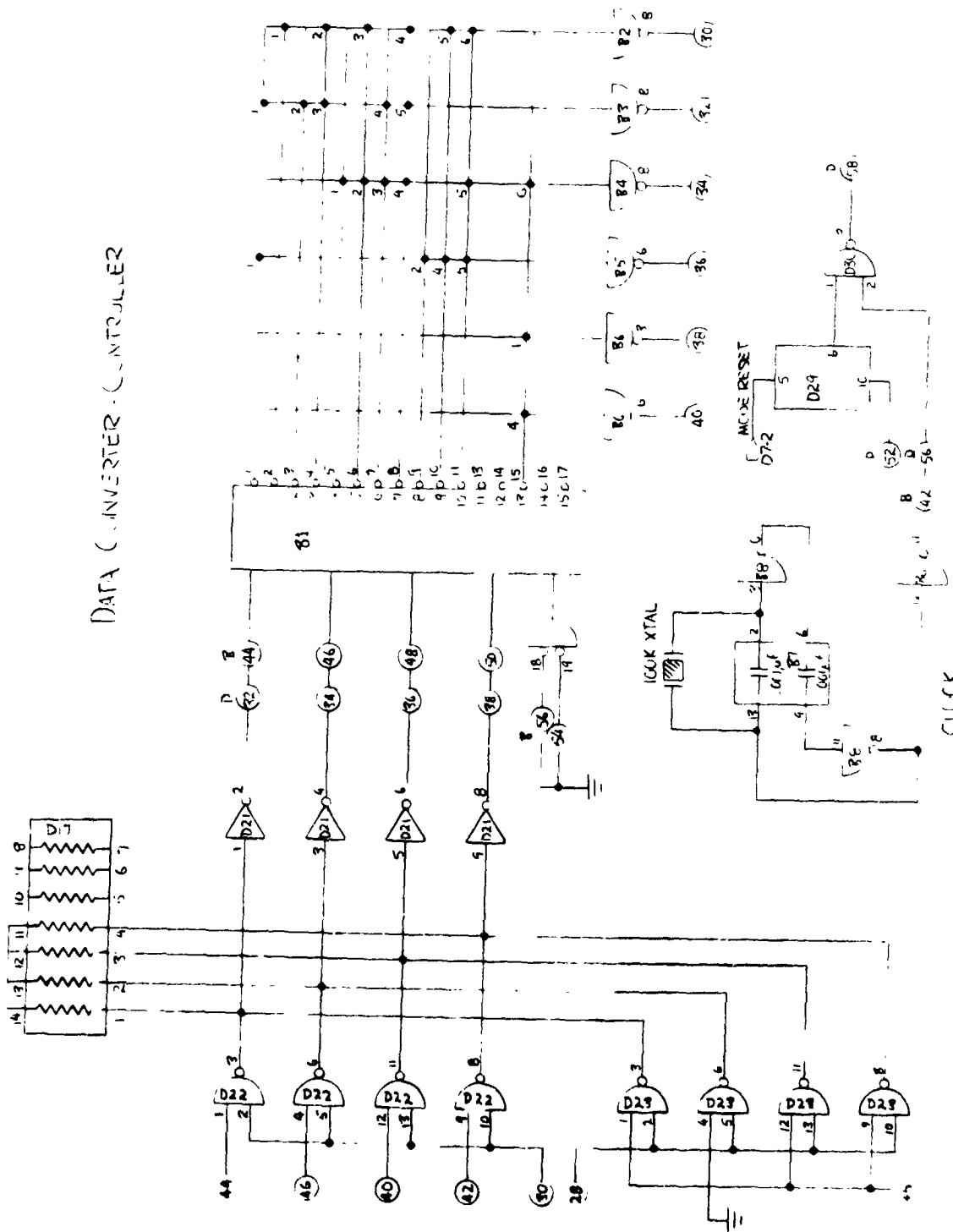


# COUNTER-SHIFT REGISTER CARD A

From	To	Function	Condition
A3	A56-C3	load shift register	input
A5	D15-C4	82s serial shift register	input
A2	B2	+ 5 volt distribution	
A4	Tens thumbwheel switch	date-1	input
A6	Tens thumbwheel switch	date-2	input
A8	Tens thumbwheel switch	date-4	input
A10	Tens thumbwheel switch	date-8	input
A12	Ones thumbwheel switch	date-1	input
A14	Ones thumbwheel switch	date-2	input
A16	Ones thumbwheel switch	date-4	input
A18	Ones thumbwheel switch	date-8	input
A20	Tens thumbwheel switch	run no. 1	input
A22	Tens thumbwheel switch	run no. 2	input
A24	Tens thumbwheel switch	run no. 4	input
A26	Tens thumbwheel switch	run no. 8	input
A28	Ones thumbwheel switch	run no. 1	input
A30	Ones thumbwheel switch	run no. 2	input
A32	Ones thumbwheel switch	run no. 4	input
A34	Ones thumbwheel switch	run no. 8	input
A36	C26	antenna identification 1	input
A38	C28	antenna identification 2	input
A40	A42-A60	antenna identification 4	ground input
A42	A40-A60	antenna identification 8	ground input
A44	D44	shift register output 1	
A46	D46	shift register output 2	
A48	D40	shift register output 4	
A50	D42	shift register output 8	
A52	C52	24s shift register mode control	input
A54	D48	real time counter reset	input
A56	A3-C3-G44	load buffer register	input
A58	D58-G56	gated oscillator 100 K Hz clock	input
A60	A40-A42-B60	common distribution	

A1	7490	decade counter
A2	7490	decade counter
A3	7490	decade counter
A4	7490	decade counter
A5	7490	decade counter
A6	7490	decade counter
A7	7495	4-bit right/left shift register
A8	7495	4-bit right/left shift register
A9	7495	4-bit right/left shift register
A10	7495	4-bit right/left shift register
A11	7495	4-bit right/left shift register
A12	7495	4-bit right/left shift register
A13	7495	4-bit right/left shift register
A14	7495	4-bit right/left shift register
A15	7495	4-bit right/left shift register
A16	7495	4-bit right/left shift register
A17	7495	4-bit right/left shift register
A18	7495	4-bit right/left shift register
A19	7495	4-bit right/left shift register
A20	7495	4-bit right/left shift register
A21	7495	4-bit right/left shift register
A22	7495	4-bit right/left shift register
A23	7495	4-bit right/left shift register
A24	7495	4-bit right/left shift register
A25	832	dual 4 input NAND gate
A26	832	dual 4 input NAND gate
A27	832	dual 4 input NAND gate
A28	7404	HEX inverter
A29	7404	HEX inverter
A30	7404	HEX inverter





# DATA CONVERTER-CONTROLLER CARD "B" AND CARD "D"

From	To	Function	Condition
B2	A2-C2	+ 5 volt distribution	
B30	F47	magnetic tape 1	output
B32	F4 <sup>c</sup>	magnetic tape 2	output
B34	F43	magnetic tape 4	output
B36	F41	magnetic tape 8	output
B38	F39	magnetic tape A	output
B40	F37	magnetic tape B	output
B42	D56	100 K Hz clock free running	output
B44	D32	shift register data 1	input
B46	D34	shift register data 2	input
B48	D36	shift register data 4	input
B50	D38	shift register data 8	input
B54	B56-B60	code conversion enable	input
B60	A60-C60	common distribution	
D28	C40	dummy data enable	input
D30	C42	real data enable	input
D32	B44	shift register data 1	output
D34	B46	shift register data 2	output
D36	B48	shift register data 4	output
D38	B50	shift register data 8	output
D40	A48	shift register data 1	input
D42	A50	shift register data 2	input
D44	A44	shift register data 4	input
D46	A46	shift register data 8	input
D52	G24	reference cavity	input
D56	B42	100 K Hz clock free running	input
D58	A58-G56	100 K Hz clock gated	output

B1	9311	one of sixteen decoder
B2	7430	8-input positive NAND gate
B3	7430	8-input positive NAND gate
B4	7430	8-input positive NAND gate
B5	7420	dual 4-input positive NAND gate
B6	7400	quad 2-input positive NAND gate
B7	2 capacitor cambion	
B8	832	dual buffer

D17	2 K resistors	
D21	7404	HEX inverter
D22	7403	quad 2-input positive NAND gate
D23	7403	quad 2-input positive NAND gate
D29	848	clocked flip-flop
D30	7400	quad 2-input positive NAND gate

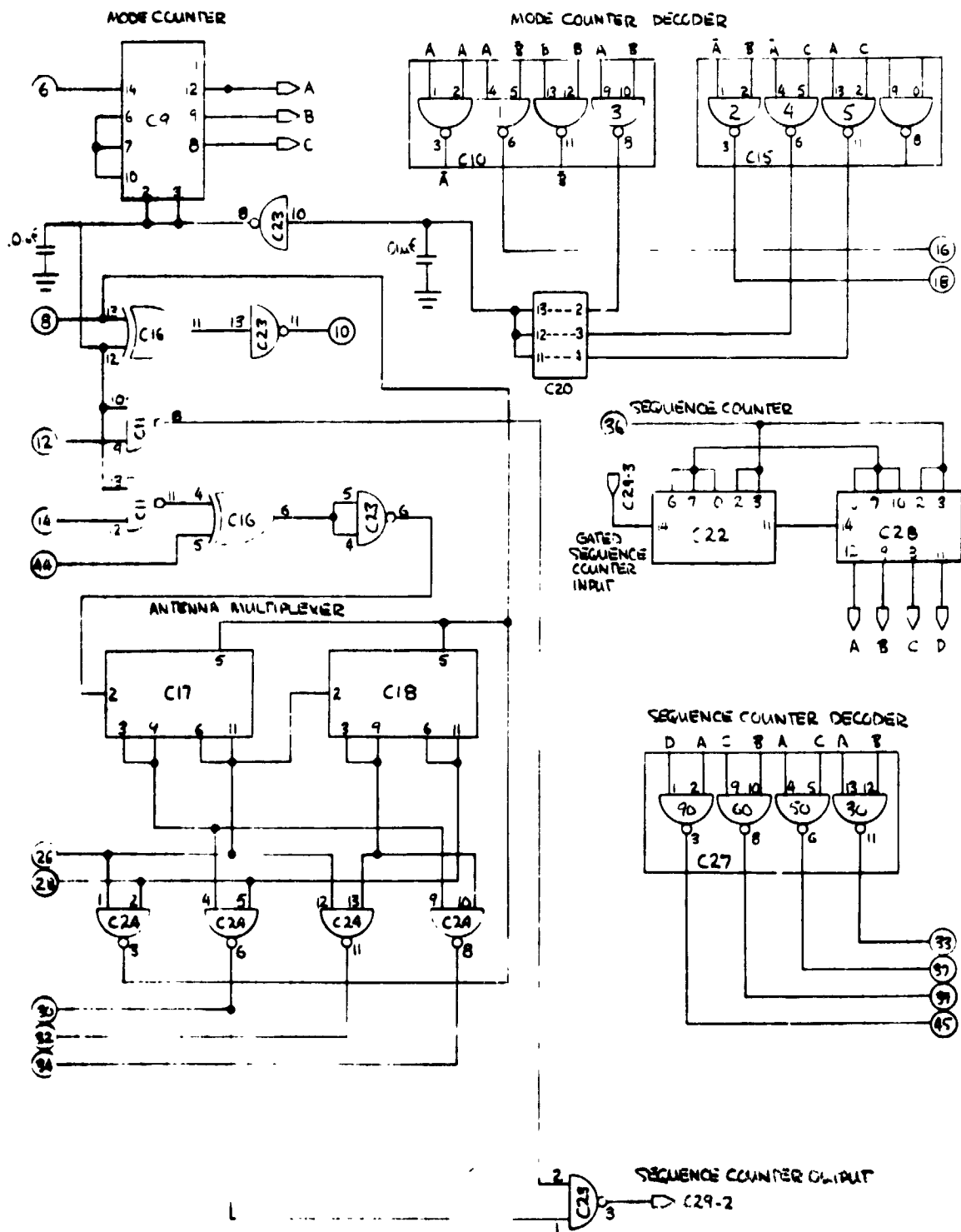


# CONTROL REGISTER-TIMING GENERATORS CARD "C"

From	To	Function	Condition
C3	A3-G44	1 $\mu$ s parallel load	output
C5	D5	serial shift gated clock	output
C49	D49-10's tws pole	sequence counter stop sweep	input
C51	D54	15 ms T.D. start sweep	input
C53	D53-N.O. SW	manual start pushbutton	input
C55	N.C. SW	manual start pushbutton	input
C57	G30	start sweep	output
C4	D15-A5	8 $\mu$ s shift clock	output
C6	G16	cryogenic cavity	input
C36	D50 N.C. SW	manual reset pushbutton	input
C40	D28	dummy data enable	output
C42	D30	real data enable	output
C52	A52	2 $\mu$ s shift register mode control	output
C58	F55	recorder clock	input

C1	7495	4-bit right/left shift register
C2	7495	4-bit right/left shift register
C3	7495	4-bit right/left shift register
C4	848	clocked flip-flop
C5	7404	HEX inverter
C6	848	clocked flip-flop
C7	RC network for one shot (C13)	
C8	RC network for one shot (C14)	
C11	7400	quad 2-input positive NAND gate
C12	848	clocked flip-flop
C13	9601	one shot multivibrator
C14	9601	one shot multivibrator
C16	7486	quad exclusive OR gate
C19	832	dual buffer
C26	848	clocked flip-flop
C29	7400	quad 2-input positive NAND gate
C30	848	clocked flip-flop

# MODE & SEQUENCE COUNTER & ANTENNA MULTIPLEXER



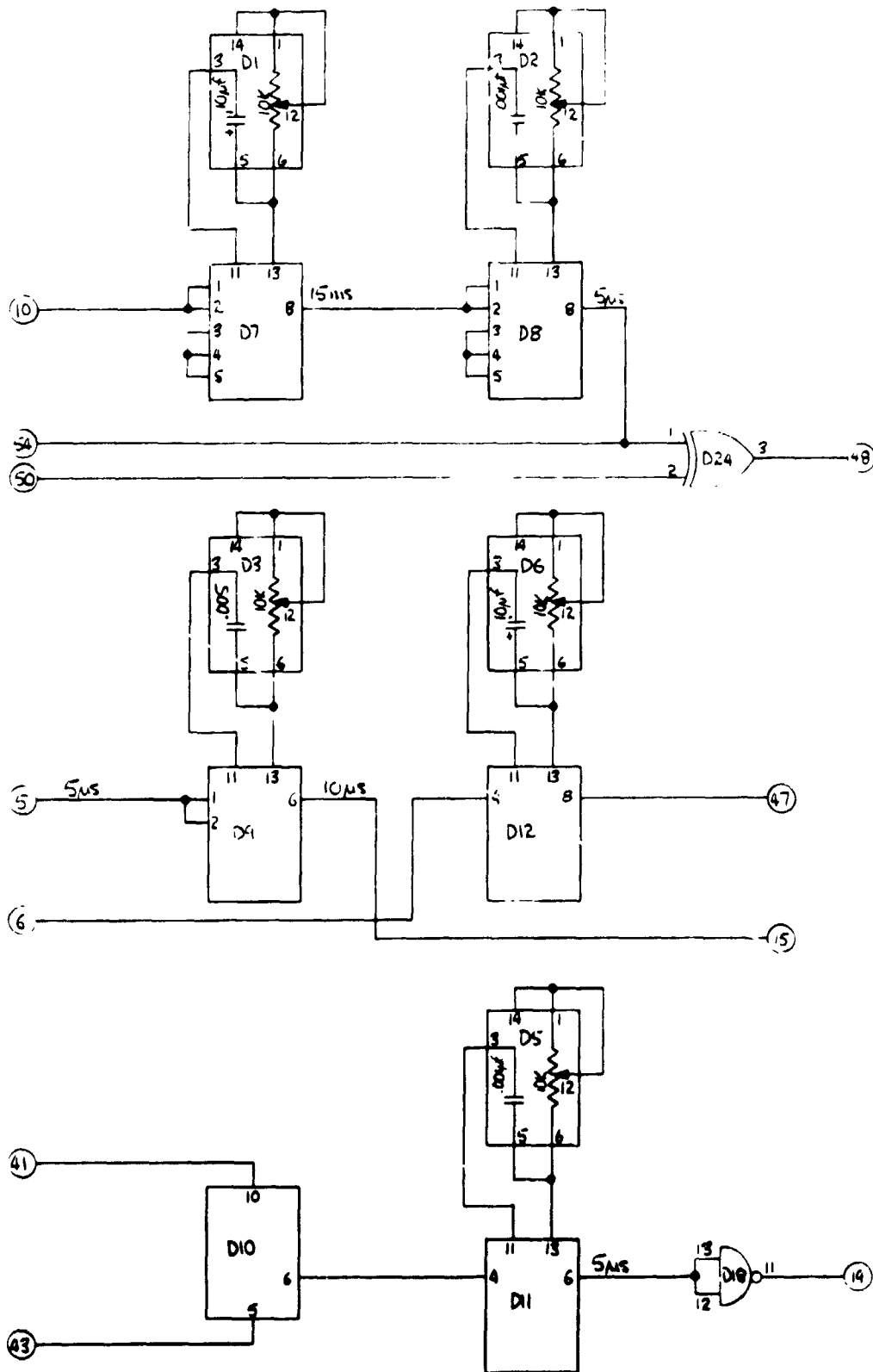
MODE COUNTER-SEQUENCE COUNTER-ANTENNA MULTIPLEXER  
CARD "C"

From	To	Function	Condition
C33	10's tws pos 3	sequence counter 30 counts	output
C37	10's tws pos 5	sequence counter 50 counts	output
C39	10's tws pos 6	sequence counter 60 counts	output
C45	10's tws pos 9	sequence counter 90 counts	output
C6	D47	cryogenic cavity gated	input
C8	N.O. sw	manual reset pushbutton	input
C10	D10-G44	reset and antenna multiplex	input
C12	N.C. sw	antenna multiplex enable	input
C14	N.O. sw	antenna multiplex enable	input
C16	G46	mode 1 reset	output
C18	G48	mode 2 reset	output
C20	G50	mode 3 reset	output
C22	G52	mode 4 reset	output
C24		mode 5 reset	output
C26	A36	antenna identification 1	output
C28	A38	antenna identification 2	output
C30	D20-E20-G40	antenna driver 1	output
C32	D16-E16-G36	antenna driver 2	output
C34	D12-E12-G32	antenna driver 4	output
C36	D50-N.C. sw	manual reset pushbutton	input
C44	D48	manual antenna advance	input



C9	7490	decade counter
C10	7400	quad 2-input NAND gate
C11	7400	quad 2-input NAND gate
C15	7400	quad 2-input NAND gate
C16	7486	quad exclusive OR gate
C17	848	clocked flip flop
C18	848	clocked flip flop
C20	cambion for wire connections	
C22	7490	decade counter
C23	7400	quad 2-input positive NAND gate
C24	7400	quad 2-input positive NAND gate
C27	846	quad 2-input NAND gate
C28	7490	decade counter

# GENERAL PURPOSE TIMING EVENTS

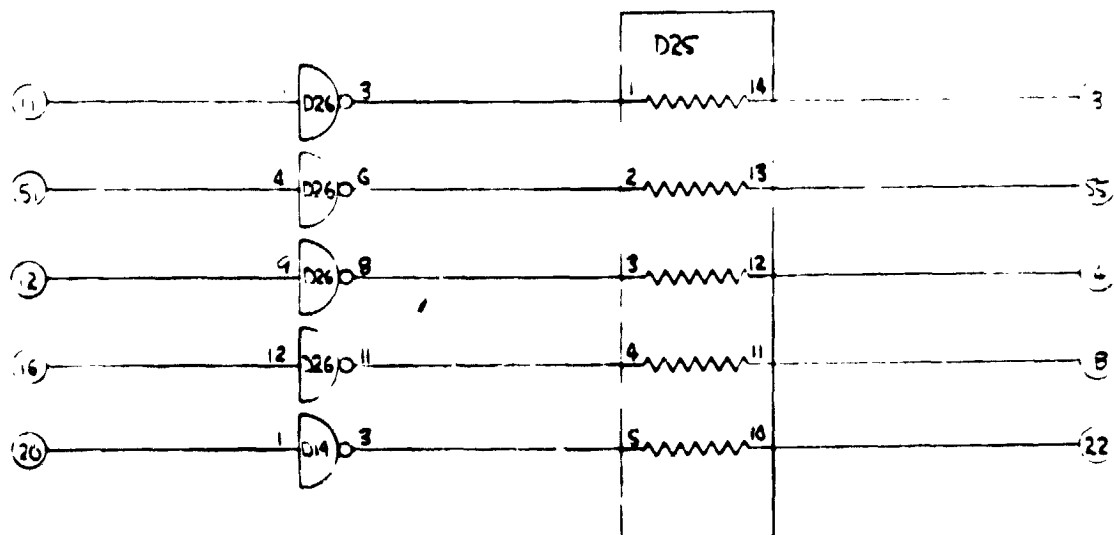
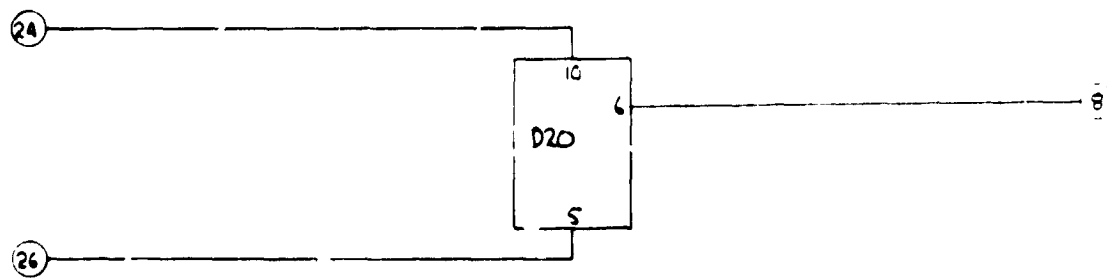
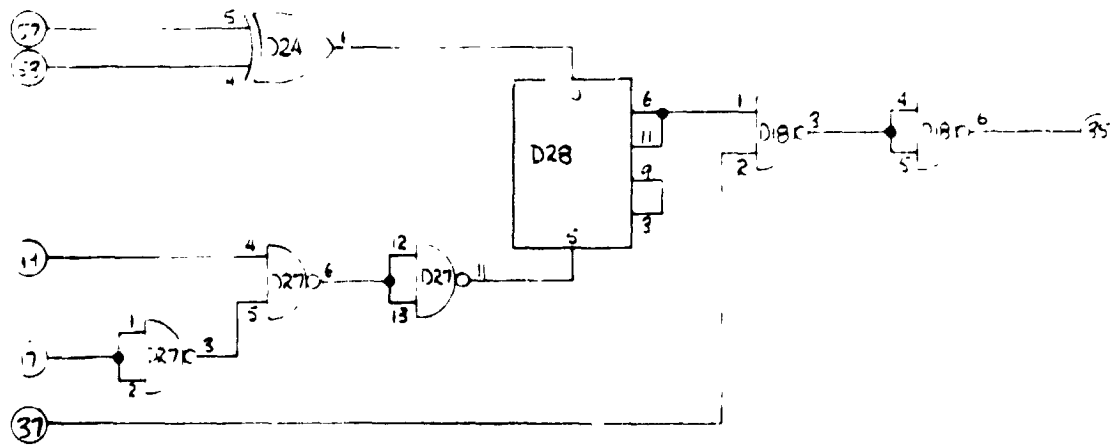


# GENERAL PURPOSE TIMING EVENTS CARD "D"

From	To	Function	Condition
D3			
D5	C5	5 $\mu$ s serial shift gated clock	input
D15	A5-C4	8 $\mu$ s serial shift gated clock	output
D19	F16	end of file 5 $\mu$ s	output
D41	N.O. sw	end of file pushbutton	input
D43	N.C. sw	end of file pushbutton	output
D47	C6	cryogenic cavity gated 5ms	output
D6	G16	cryogenic cavity gated	input
D10	C10	reset, antenna multiplex and 15ms TD	input
D48	A54-G54	real time counter reset	output
D50	C36-N.C. sw	manual reset pushbutton	input
D54	C51	15ms time delay	output

D1	RC network for one shot (10 $\mu$ f, 10 K)
D2	RC network for one shot (.001 $\mu$ f, 10 K)
D3	RC network for one shot (.005 $\mu$ f, 10 K)
D5	RC network for one shot (.001 $\mu$ f, 10 K)
D6	RC network for one shot (10 $\mu$ f, 10 K)
D7	9601 one shot multivibrator
D8	9601 one shot multivibrator
D9	9601 one shot multivibrator
D10	848 clocked flip-flop
D11	9601 one shot multivibrator
D12	9601 one shot multivibrato
D18	7400 quad 2-input positive AND gate
D24	7486 quad 2-input exclusive OR gate

# GENERAL PURPOSE FUNCTIONS

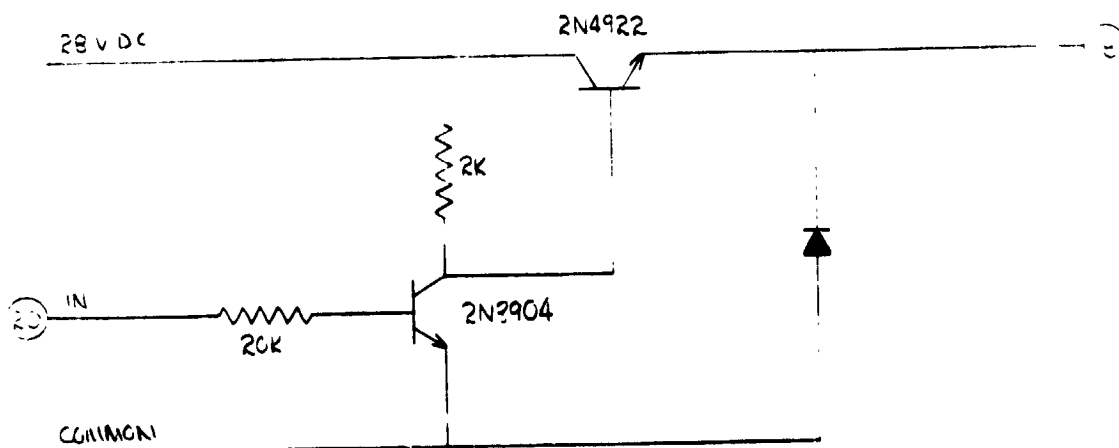
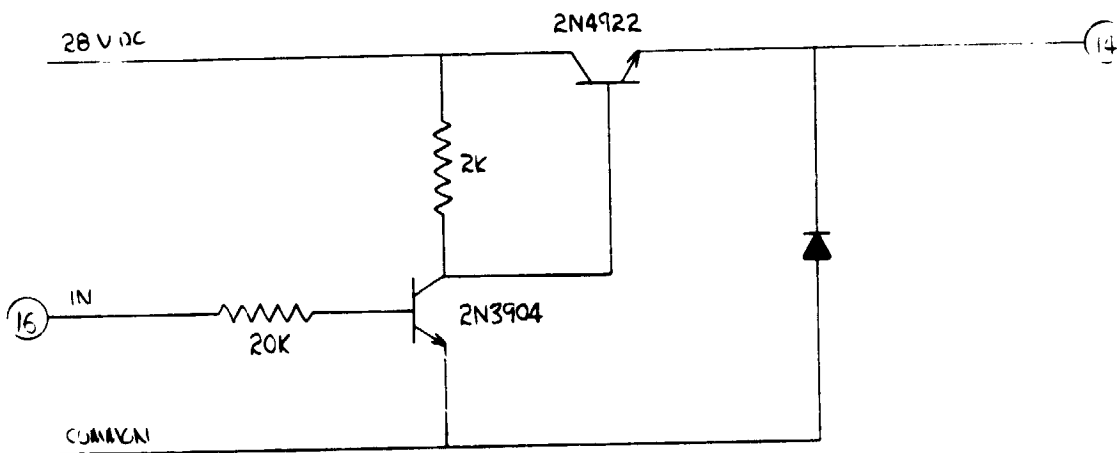
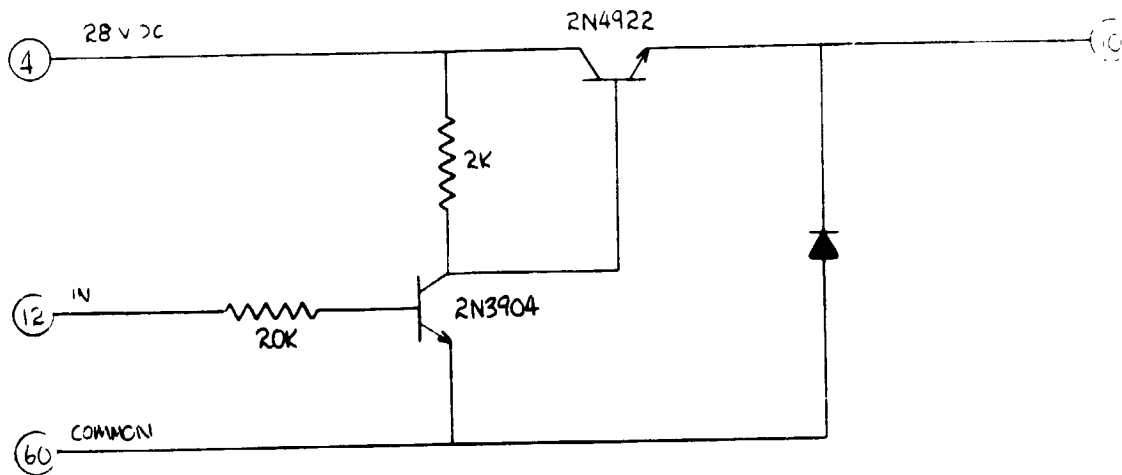


# GENERAL PURPOSE FUNCTIONS CARD "D"

From	To	Function	Condition
D11	F11	tape running	input
D13	led 1 panel	tape running indicator	output
D17	F15	end of tape	input
D35	F35	write command	output
D37	F55-C58	write shift clock	input
D49	C49 10's tws	sequence counter switch pole	input
D51	F53	file protect missing	input
D53	C53-N.O. sw	manual start pushbutton	input
D55	led 2 panel	file protect missing indicator	output
D57	N.O. sw	recorder on clamp	input
D8	C44	manual advance antenna	output
D12	E12-G32-C34	antenna "0"	input
D14	led ant "0"	antenna "0" indicator	output
D16	E16-G36-C32	antenna "1"	input
D18	led ant "1"	antenna "1" indicator	output
D20	E20-G40-C30	antenna "2"	input
D22	led ant "2"	antenna "2" indicator	output
D24	N.O. sw	manual antenna advance	input
D26	N.C. sw	manual antenna advance	input

D18	7400	quad 2-input positive NAND gate
D19	7403	quad 2-input positive NAND gate
D20	848	clocked flip-flop
D24	7486	quad 2-input exclusive OR gate
D25	510 $\Omega$ resistors	
D26	7403	quad 2-input positive NAND gate
D27	7400	quad 2-input positive NAND gate
D28	848	clocked flip-flop

# ANTENNA SWITCH DRIVERS





# ANTENNA SWITCH DRIVERS CARD "E"

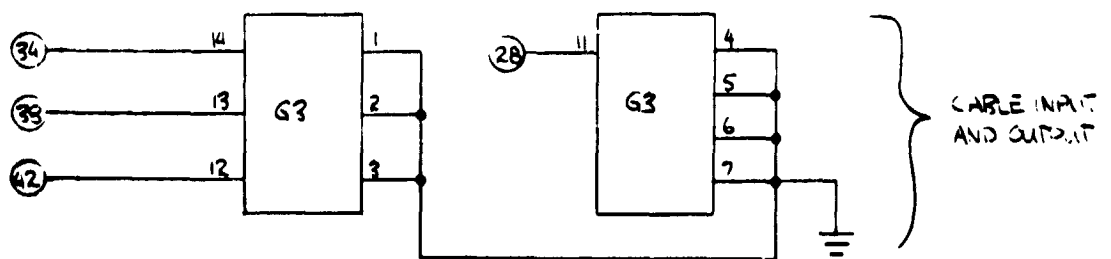
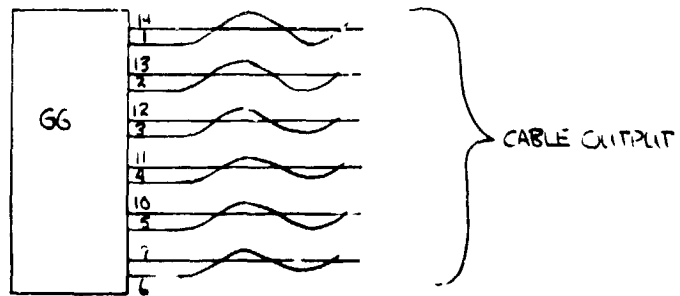
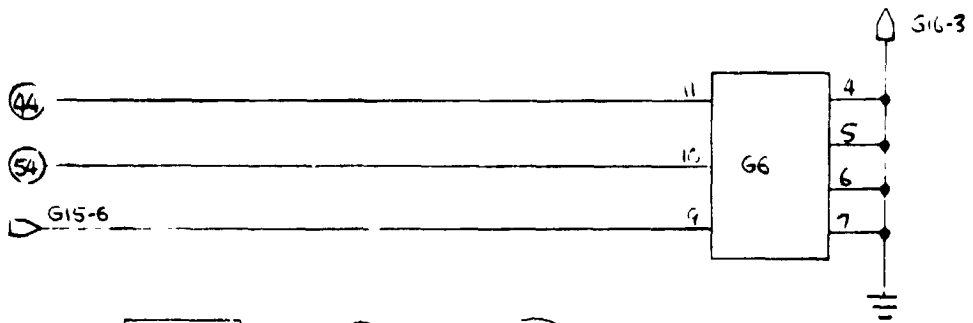
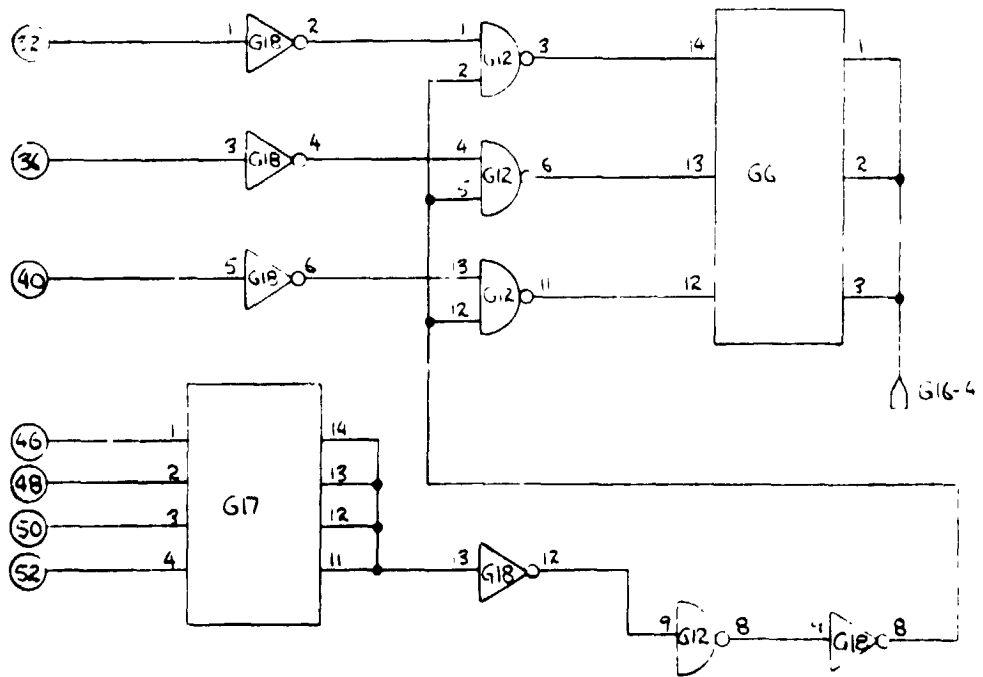
From	To	Function	Condition
E4	G28	+ 28 volts	input
E10	G34	antenna "0"	output
E12	C34-D12-G32	antenna "0"	input
E14	G38	antenna "1"	output
E16	C32-D16-G36	antenna "1"	input
E18	G42	antenna "2"	output
E20	C30-D20-G40	antenna "2"	input



CARD "G"

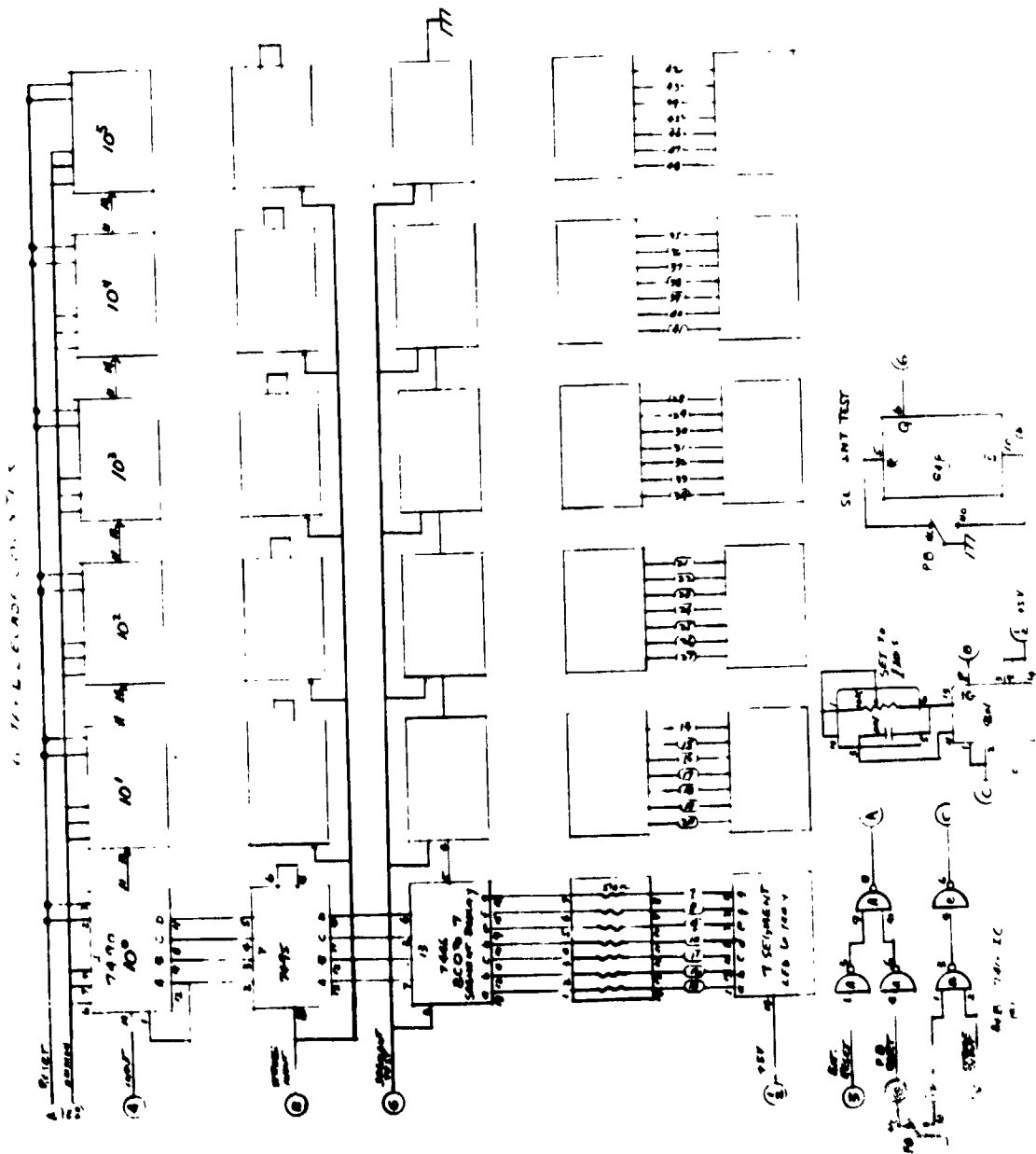
From	To	Function	Condition
G6	connector 1	cryogenic cavity signal	input
G8		amplifier "1"	test point
G10		amplifier "2"	test point
G14		amplifier "3"	test point
G16	D6	cryogenic signal gated	output
G18		cryogenic comparator signal	test point
G20	connector 2	reference cavity signal	input
G22		reference cavity amplifier "1"	test point
G24	D52	reference cavity comparator	output
G54	D10-C10	mode counter reset signal	input
G56	D58-A58	100 K Hz clock	input
G58		reference cavity amplifier "2"	test point

G1	LM301 operational amplifier	
G2	1M $\Omega$ pot and 1M $\Omega$ resistor	
G4	LM301 operational amplifier	
G5	1M $\Omega$ pot and 1M $\Omega$ resistor	
G7	LM301 operational amplifier	
G8	capacitor cambion	
G9	848	clocked flip-flop
G10	LM301 operational amplifier	
G11	capacitor cambion	
G13	LM301 operational amplifier	
G14	LM301 operational amplifier	
G15	7400	quad 2-input NAND gate
G20	848	clocked flip-flop



From	To	Function	Condition
G28	E4	+ 28 volt	output
G32	D12-E12-C34	antenna "0"	input
G34	E10	antenna "0" drive	input
G36	D16-E16-C32	antenna "1"	input
G38	E14	antenna "1" drive	input
G40	D20-E20-C30	antenna "2"	input
G42	E18	antenna "2" drive	input
G44	C3-A3	parallel load	input
G46	C16	mode 1 counter	input
G48	C18	mode 2 counter	input
G50	C20	mode 3 counter	input
G52	C22	mode 4 counter	input
G54	D48-A54	15 ms time delay reset	input

G3	cambion for wire connections	
G6	cable cambion	
G12	7400	quad 2-input positive NAND gate
G17	cambion for wire connections	
G18	7404	HEX inverter





### 6 STAGE DECADE COUNTER CARD

2/28/73	Connector Pins	Function
	1	+ 5V
	2	+ 5V
	3	external reset
	4	signal input
	5	strobe input
	6	segment test
	56	push button reset (N. O. )
	57	
	58	push button reset (N. C. )
	59	common
	60	common

# INTERFACE CARD

2/28 /73	Connector Pins	Function
	1	+ 5V
	2	+ 5V
	3	100 KHz clock
	4	common
	5	
	6	
	7	
	8	
	9	strobe #1 ANT "0"
	10	common
	11	strobe #2 ANT "1"
	12	common
	13	strobe #3 ANT "2"
	14	common
	15	external reset (all resets)
	16	common
	17	
	18	
	19	parallel load
	20	common
	59	common
	60	common

Table E-1. Interface Connector Pin List

ACT. PIN	SIG. GRD.	SIGNAL (see notes 1 through 3)	ACTIVE WIRE COLOR	7-TRACK	9-TRACK	BASIC	WRITE ONLY	READ ONLY	READ WRITE	RAW	SPECIAL
-------------	--------------	--------------------------------	-------------------------	---------	---------	-------	------------	-----------	------------	-----	---------

## TRANSPORT CONTROL SIGNALS

3		Run Normal Speed	input level	80	X	X	X	X	X	X	
5		Run High Speed	input level	30	X	X	X	X	X	X	
7		Forward Select	input level	70	X	X	X	X	X	X	
9		Reverse Select	input level	60	X	X	X	X	X	X	
11		Tape Running	output level	50	X	X	X	X	X	X	
13		Load Point	output level	8	X	X	X	X	X	X	
15		End of Tape	output level	6	X	X	X	X	X	X	
17		Broken Tape	output level	5	X	X	X	X	X	X	
19		Unload Command	input pulse	10	X	X	X	X	X	X	
21		Rewind to Load Point	input pulse	90	X	X	X	X	X	X	
23		Off Line	input pulse	9	X	X	X	X	X	X	
25		Rewind in Process	output level	20	X	X	X	X	X	X	
27		On Line	output level	40	X	X	X	X	X	X	
		Reserved - Do Not Use		-							X
		Reserved - Do Not Use		0							X
29		Chassis Ground (see note 4)		1	X	X	X	X	X	X	

## WRITE SIGNALS

31		Write Select	input level	7	X	X	X			X	
33		Write Parity Select (odd)	input level	92	X			X		X	
		Reserved		0	X	X					X
35		Write Command	input pulse	50	X	X		X		X	
18		Data Channel 0 (Write)	input level	97		X		X		X	
20		Data Channel 1 (Write)	input level	90		X		X		X	
37		Data Channel 2-B (Write)	input level	93	X	X		X		X	
39		Data Channel 3-A (Write)	input level	96	X	X		X		X	
41		Data Channel 4-8 (Write)	input level	95	X	X		X		X	
43		Data Channel 5-4 (Write)	input level	94	X	X		X		X	
45		Data Channel 6-2 (Write)	input level	9	X	X		X		X	
47		Data Channel 7-1 (Write)	input level	98	X	X		X		X	
49		Data Channel P-C (Write)	input level	30	X	X					X
51		Write Status	output level	90	X	X		X		X	
53		File Protect Ring Missing	output level	98	X	X		X		X	
55		Write Clock	output pulse	70	X	X		X		X	
57		Write Clock Gate	output level	60	X	X		X		X	
22		Write Echo Error	output pulse	10	X	X		X			
12		2X Write Clock	input pulse	5	X	X					X
14		Write LRC	input pulse	20	X			X		X	
16		EOF Command	input pulse	80	X	X		X		X	

## APPENDIX C

### Q MEASUREMENT SUMMARY

The objective of the Q measurement task was to determine the spherical vessel Q as a function of diameter and wall material. The results of this measurement task are summarized in tabular form.

<u>Spherical Vessels</u>			
<u>Mode</u>	<u>19" Cu</u>	<u>18" SS</u>	<u>60" SS</u>
TM <sub>11</sub>	91,000	11,600	6,200
TM <sub>21</sub>	85,200	10,900	13,600
TE <sub>11</sub>	81,100	19,700	28,400
TM <sub>21</sub>	41,200	10,900	17,500

Considering the lowest Q (TM<sub>11</sub> in the 60" SS),  $f_1$  is 172 MHz when empty and would be 155 MHz when filled with boiling point LH<sub>2</sub>. The total frequency shift from empty to full would be 17 MHz. The bandwidth at the 3dB points is 0.028 MHz, or approximately 0.15% of the empty-full bandwidth. For LOX the 3dB bandwidth is approximately 0.10% of the empty-full bandwidth. Thus it appears that the Q will not seriously degrade the RF resonance technique in stainless steel spherical vessels up to 5 feet in diameter.

Let us assume that the change in Q pattern measured with the 18" and 60" stainless steel vessels continues to even larger diameters. Then for a 15 foot diameter stainless steel vessel, the 3dB bandwidth is 0.3% of the empty full LH<sub>2</sub> bandwidth. The LOX 3dB bandwidth is approximately 0.2% of the empty-full bandwidth.

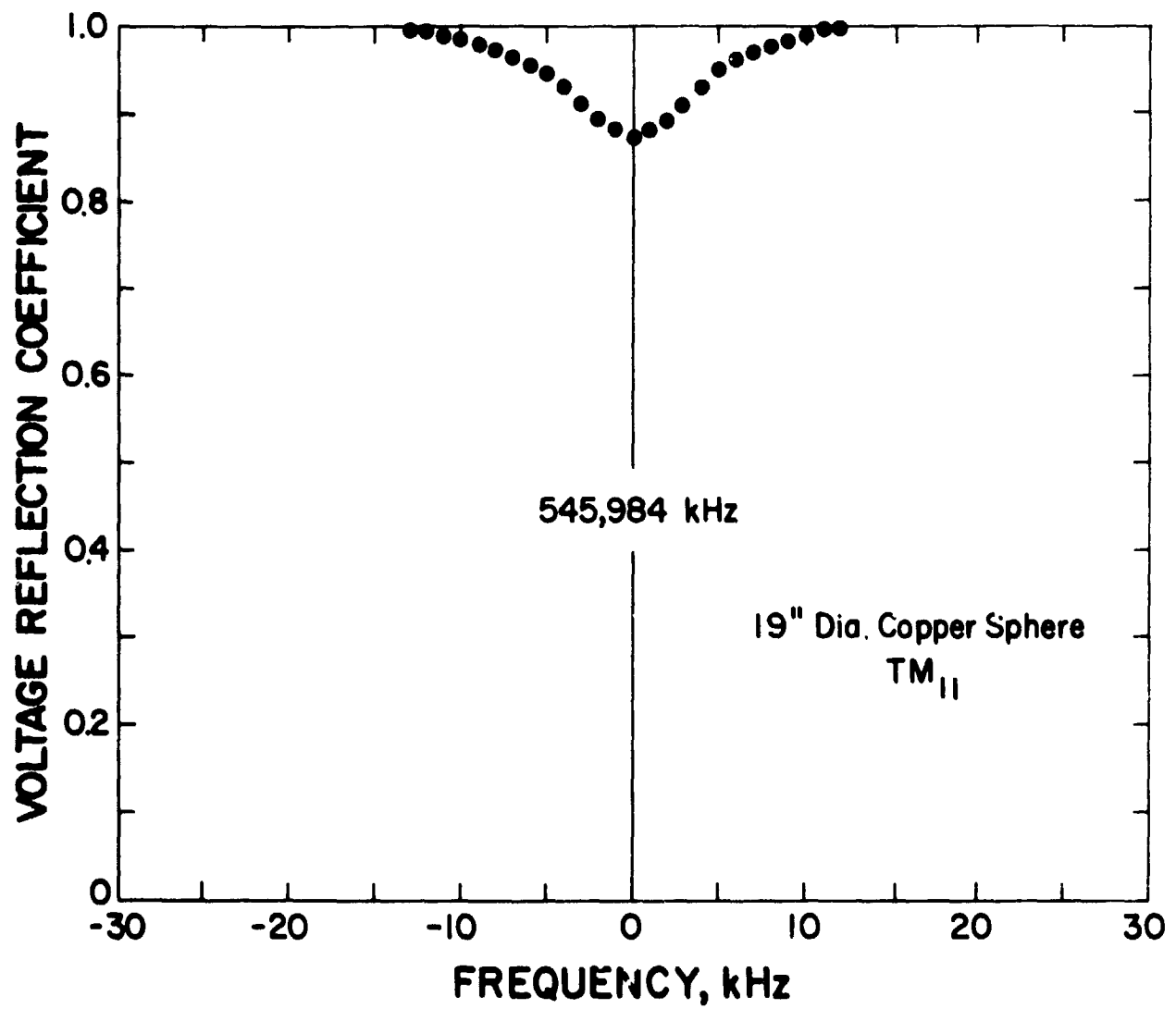


Figure C1. Frequency response of the copper spherical vessel for the  $TM_{011}$  mode.

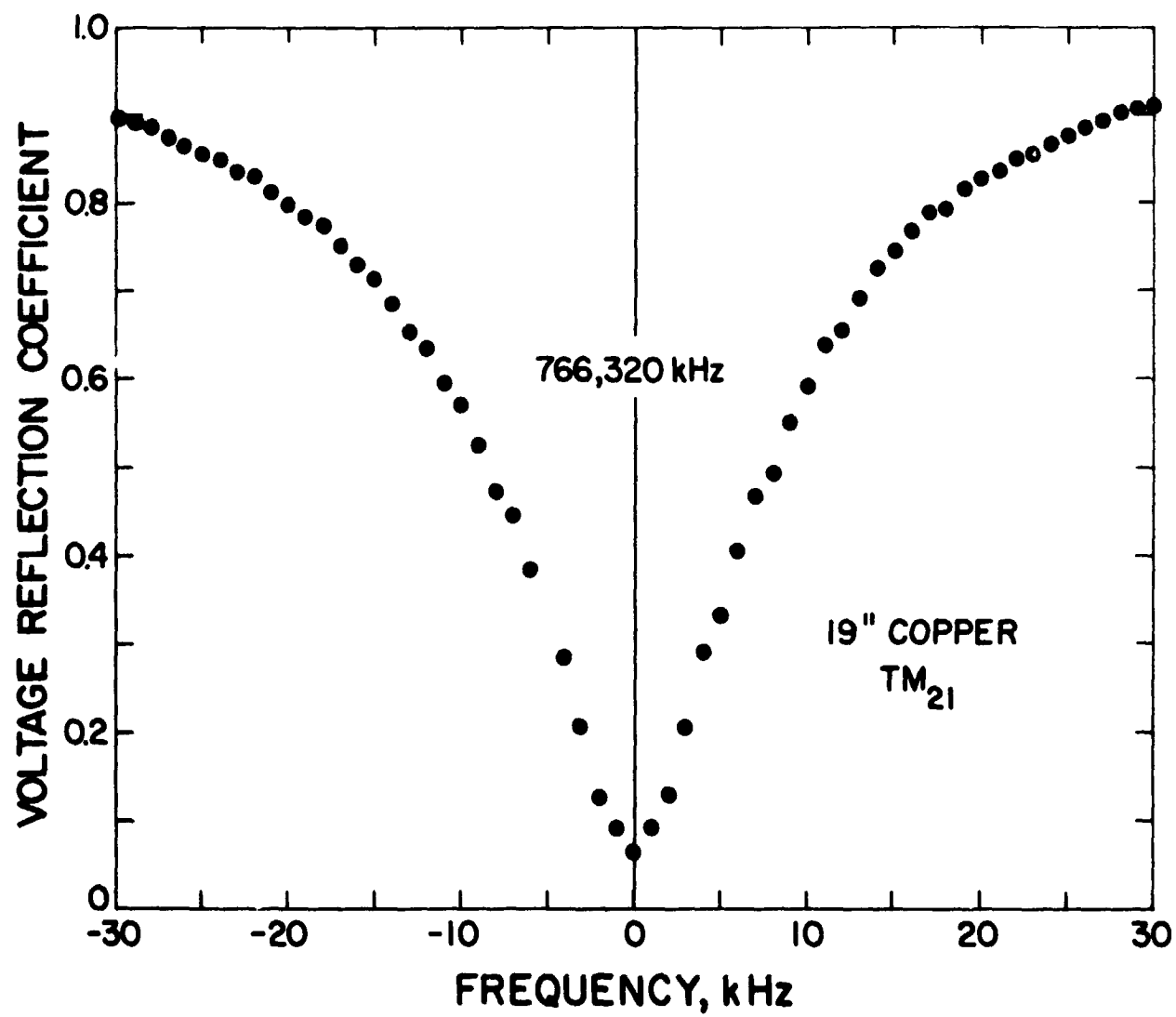


Figure C2. Frequency response of the copper spherical vessel for the  $TM_{021}$  mode.

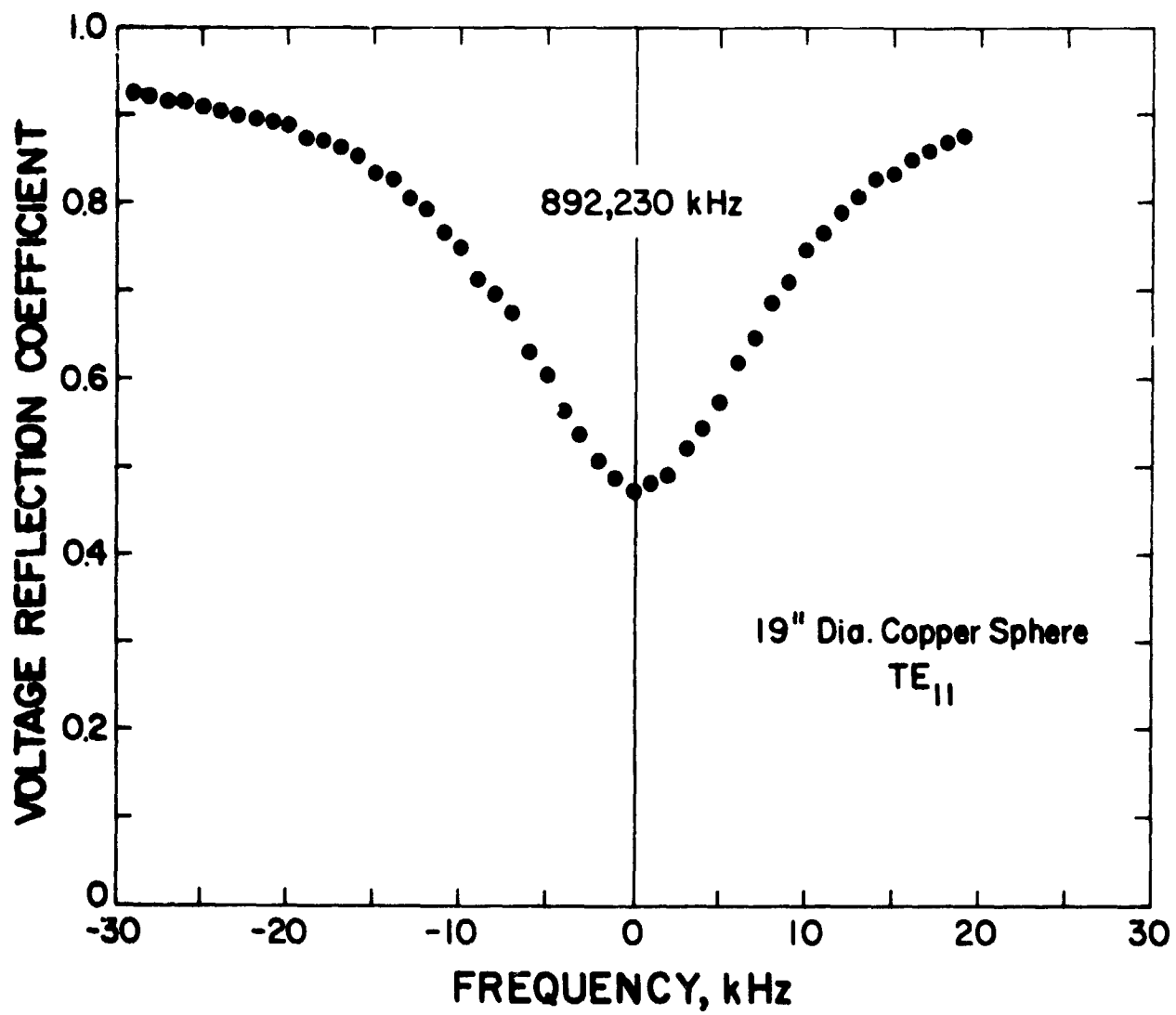


Figure C3. Frequency response of the copper spherical vessel for the  $TE_{11}$  mode.

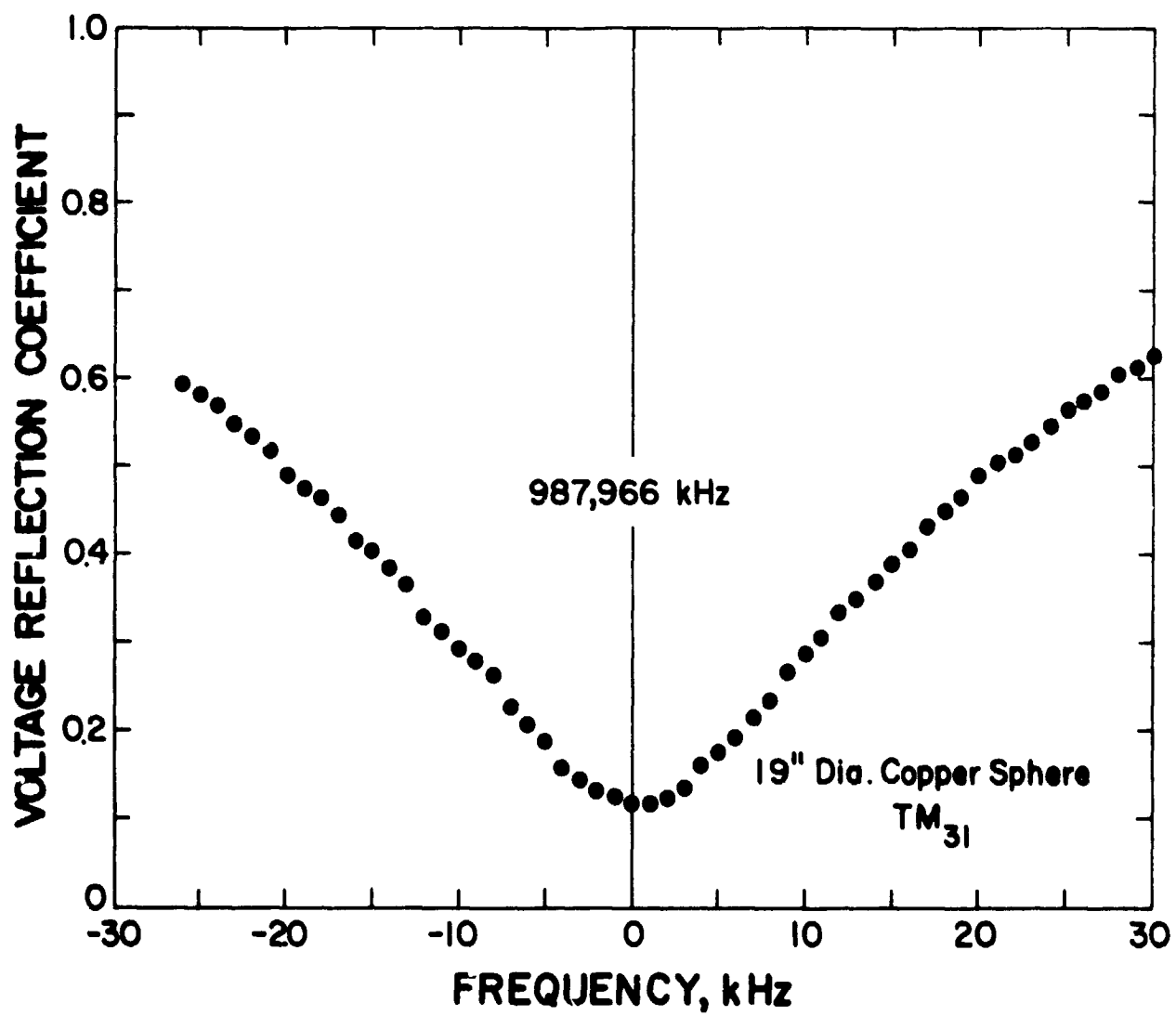


Figure C4. Frequency response of the copper spherical vessel for the  $TM_{031}$  mode.



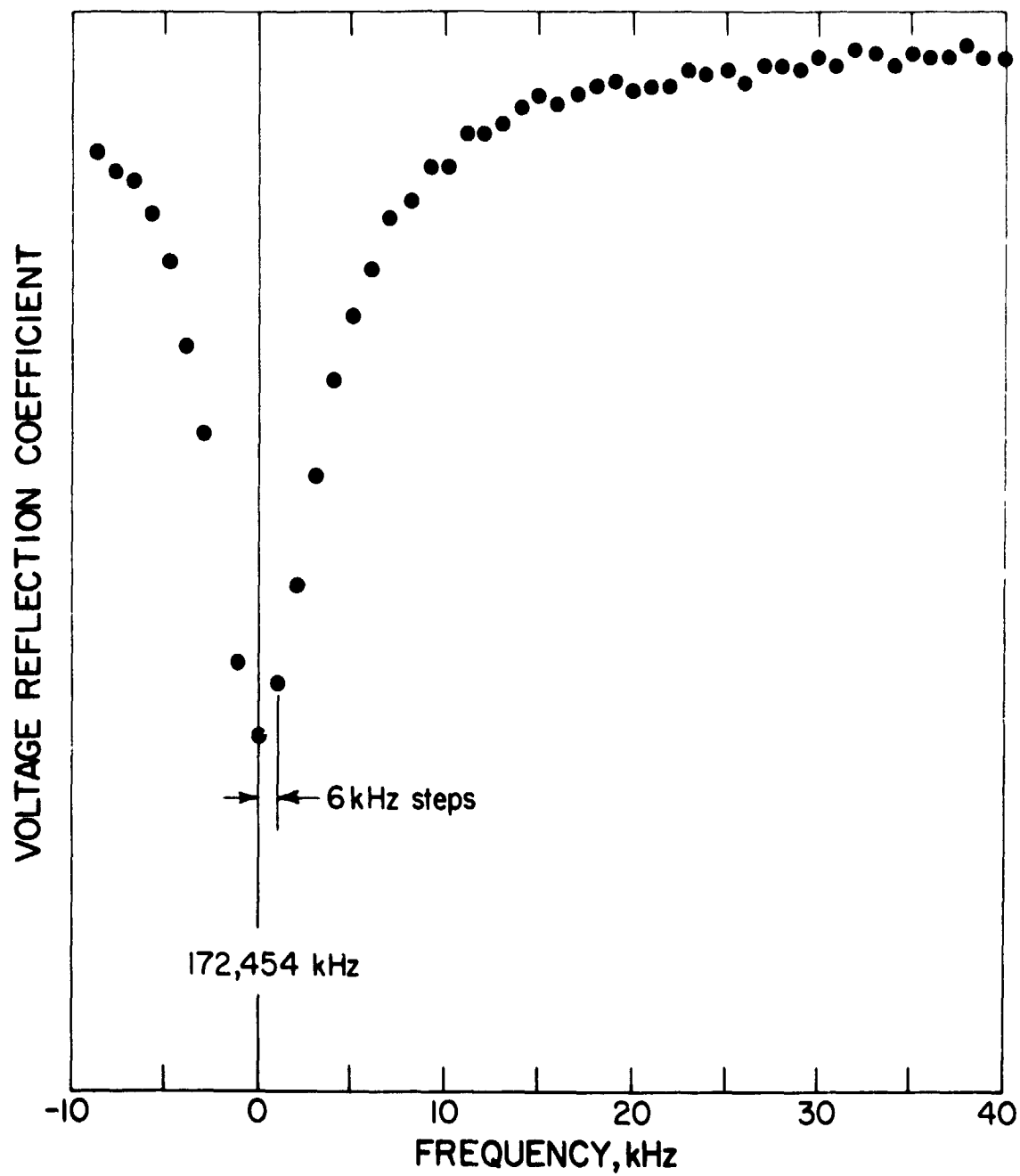


Figure C5. Frequency response of the 60 inch diameter stainless steel sphere -  $TM_{011}$  mode.

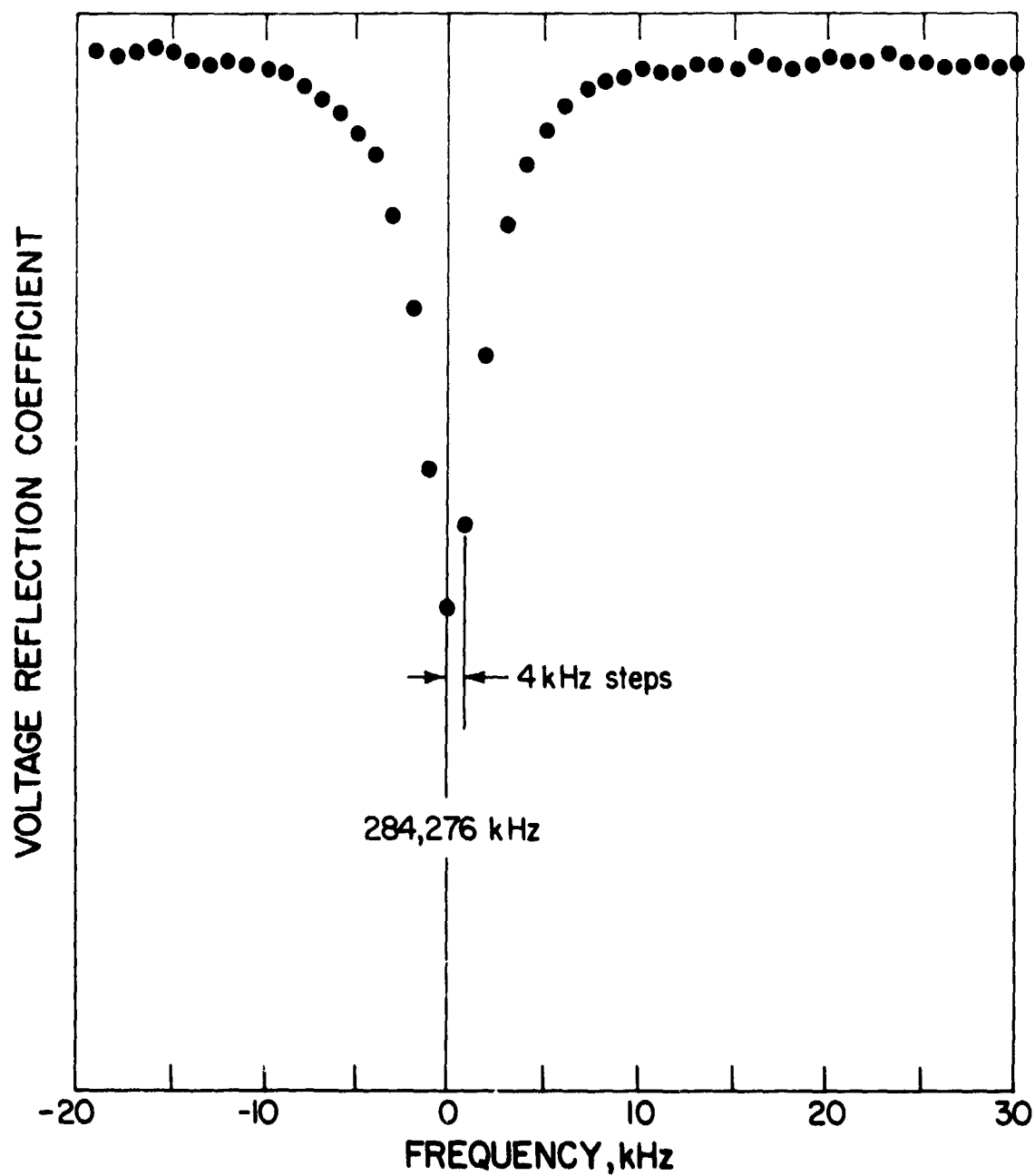


Figure C6. Frequency response of the 60 inch diameter stainless steel sphere -  $TE_{011}$  mode.

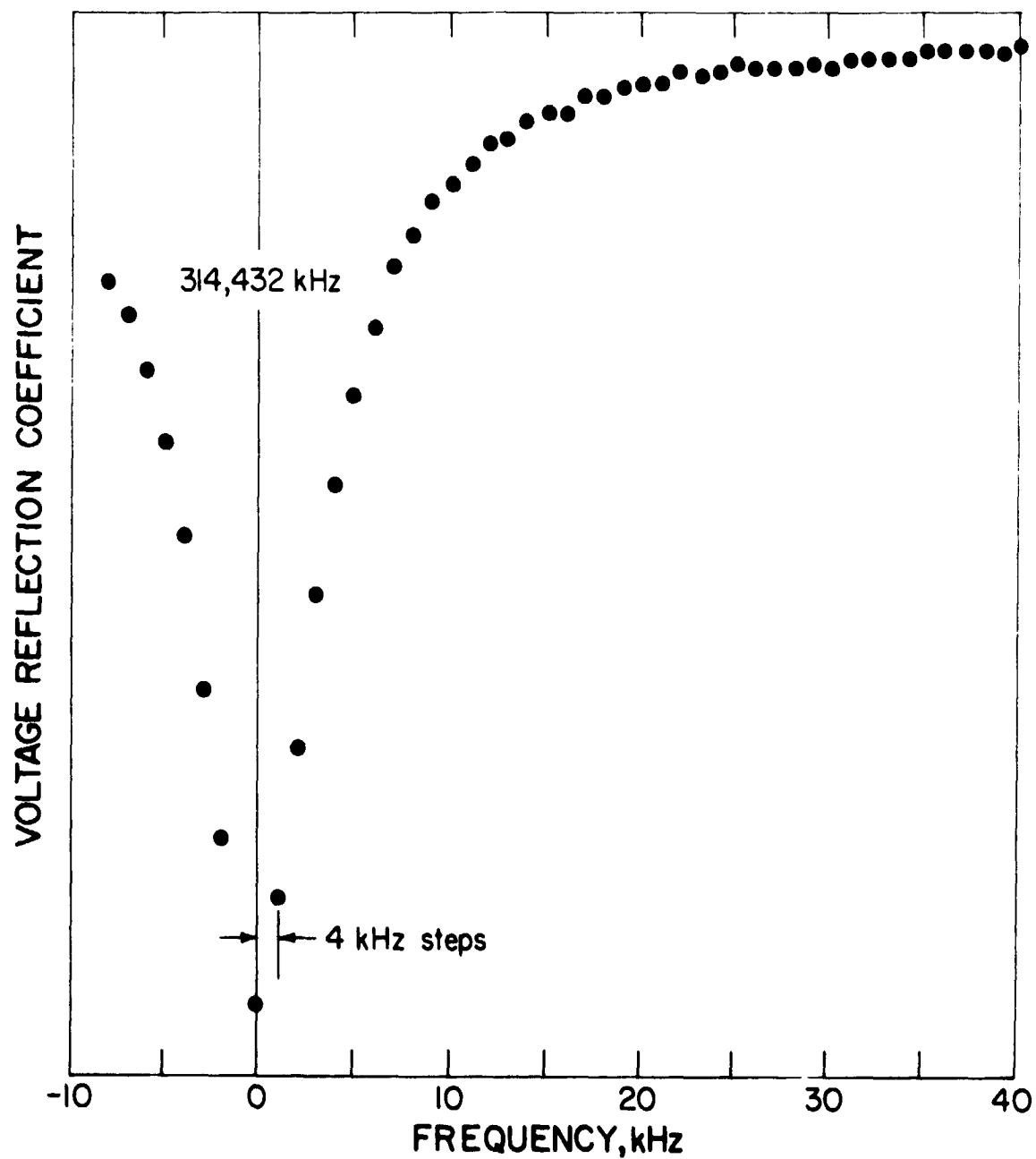


Figure C7. Frequency response of the 60 inch diameter stainless steel sphere -  $TM_{031}$  mode.

## APPENDIX D

### DATA REDUCTION FROM THE MAGNETIC TAPE UNIT

The magnetic tape unit stores the information coming from the data acquisition system described in Appendix B. Each data run lasts about ten seconds; during this time the system records about 25 independent measurements for each mode or about 375 data points in all (counting 3 antennas and 5 modes for each antenna). The data reduction processes the raw tape, finds the average and standard deviation of the 25 data points for each mode and punches these numbers on to standard data cards along with other identification codes. Data from other sources are also punched on these cards i. e. , mass, pressure and temperatures. The cards are then ready for plotting routines, and other types of data analysis.

#### I. Program DMPMODES

Since the tape recorder in the laboratory is a continuous write recorder and not a sequential recorder, the letter D is hard-wired to the output whenever there is no data being recorded. Most of the tape is, in fact, dummy data. DMPMODES accomplishes many functions. It takes the data from the magnetic tape made in the laboratory, deletes all D's and other non-essential characters, sorts the 12 character data by antenna, lists this data and writes this data on a second magnetic tape.

Essentially, the program works as follows. A tape record is read into array M. A pointer, KNT, moves down the array. If the word in M(KNT) is all D's, KNT is incremented. When a word that is not all D's comes along, the character: (or 15 octal) is searched for. When a: is found, it and the characters surrounding it are placed into the first two locations of the array IW. Thinking of these two 8 byte words as one 16 byte word, it should look like  $d_1 d_2 r_1 r_2 a: t_1 t_2 t_3 t_4 t_5 t_6$

DDDD, where all subscripted letters are integers, and D's are dummy data.  $d_1 d_2$  is the data code selected from the recorder front panel.  $r_1 r_2$  is the run code, also selected from the front panel.  $a_1$  is the antenna code.  $t_1-t_6$  are the data. If this 16 byte "word" is not correct, that is, if an alpha character appears where an integer is supposed to be, the data point is rejected and the program starts searching for the next data point. If the data point appears to be all right, it is decoded, and the data is printed under the appropriate antenna column. Also, the 16 byte "word" is written on a second magnetic tape, thus saving later programs from having to re-sort the mass of characters on the input tape.

## II. Program AVMODES

Basically, this program takes data from the second magnetic tape, sorts it by antenna and mode, finds the average time and standard deviation for each mode for each antenna, lists these averages and standard deviations, and punches on standard data cards the averages and standard deviations, along with the date, date code, run code, antenna number, mode number, and the number of points used to calculate the average and standard deviation. The input magnetic tape is essentially a sequence of the magnetic tapes written by DMPMODES. The tape has all the data from all the days of running the experiment. Each day is separated from the others by an end of file marker, and each run is separated from other runs by an inter-record gap. The first record of each file contains the date of the run.

The program works as follows. The first record is read into the first 10 words of array M. The date of the run is extracted from these ten words and stored into the variable IDATE. Each of the other records in the file are read into the array M, one at a time. Every

two words of the array M are the same as in the 16 byte "word" described above in the discussion of DMPMODES. This 16 byte word is decoded, or broken up, into the date code, IDC; the run code, IRC; the antenna code, IA; the colon character, ICOL; and the time data, ITIME. The time is checked to see which mode, if any, it belongs to. The point is then used to calculate the average and standard deviation of that particular mode of that particular antenna. Note that for each record on the tape, or run, there is a possible total of 15 of these averages and standard deviations to be printed and punched (corresponding to three antennas with five modes each). The punch card output is used for graph routines and other data analysis programs.

III. The following pages give a listing of these programs which are being used on the CDC 3800 computer.

```

PROGRAM DMPMODES
DIMENSION MSG(3),IW(8),IHZ(3),IHZSQ(3),T(3),TSQ(3),K(3),IOA(5000)
COMMON M(32765)
1 FORMAT(I5)
2 FORMAT(*1RECORD *,I5,* LONGER THAN 32764, SOME LOST.*)
3 FORMAT(8R1)
4 FORMAT(2I2,I1,R1,I6,4X)
5 FORMAT(*1*,57X,*TAPE RECORD NUMBER *,I3,3A8/*8*
151X,*DATE CODE9 *,I2,5X,*PUN CODE9 *,I2/
255X,*ANTENNA IDENTIFICATION CODE*/33X,*0*,32X,*1*,32X,*2*/)
6 FORMAT(31X,I6)
7 FORMAT(64X,I6)
8 FORMAT(97X,I6)
9 FORMAT(*0AVG*,27X,F9.2,2(23X,F9.2)/ * SIGMA*,23X,E12.6,2(21X,E12.6)
1/1X,I4,* SCANS DELETED IN THIS RECORD.*)
NRUNO=999
CALL 09OVER
CALL 10HCHECK
NREC=0
READ 1,NSKIP
IF(EOF,60)10,103
103 DO 105 I=1,NSKIP
BUFFERIN(1,0)(M,M)
104 IF(UNIT,1)104,105
105 CONTINUE
10 BUFFERIN(1,0)(M,M(32765))
IF(NREC.EQ.0.OR.KCA.EQ.1)GO TO 115
BUFFEROUT(2,1)(IOA,IOA(KOA-1))
115 KOA=1
MSG(1)=MSG(2)=MSG(3)=8H
KNT=0
IHZ(1)=IHZ(2)=IHZ(3)=0
K(1)=K(2)=K(3)=0
IHZSQ(1)=IHZSQ(2)=IHZSQ(3)=0
NREC=NRFC+1
IDROP=-1
LINE=75
KCODE=0
117 IF(UNIT,2)117,11
11 IF(UNIT,1)11,15,99,13
13 MSG(1)=8H CONTAIN
MSG(2)=8H5 PARITY
MSG(3)=8H ERRORS
15 L=LENGTH(HF(1))
IF(L.GT.4)GO TO 195
NREC=NREC-1
GO TO 10
193 IF(L.LT.32764)GO TO 198
PRINT 2,NREC
198 IDROP=IDROP+1
205 KNT=KNT+1

```

```

        IF(KCODE.EQ.1)KNT=KNT+1
        KCODE=0
20  IF(KNT.GT.L-2)23,24
23  DO 242 I=1,3
        FI=K(I)
        T(I)=IHZ(I)
        T(I)=T(I)/FI
        IF(K(I).GT.1)GO TO 241
        TSQ(I)=0.
        GO TO 242
241 TSQ(I)=IHZSQ(I)
        TSQ(I)=SQRT(ABS((FI*TSQ(I)-T(I)**2)/FI/(FI-1.)))
242 CONTINUE
        PRINT 9,(T(I),I=1,3),(TSQ(I),I=1,3),IDROP
        GO TO 10
24  IF(M(KNT).EQ.8HDDDDDDDD)GO TO 27
        DECODE(8,3,M(KNT))IW
25  DO 26 I=1,8
        IF(IW(I).NE.15B)GO TO 26
        NSH=I-6
        IF(NSH)29,37,30
26  CONTINUE
27  KNT=KNT+1
        GO TO 20
29  KNT=KNT-1
        NSH=NSH+8
        KCODE=1
30  MSK1=2**(6*NSH)
        MMN1=MSK1-1
        MSK2=2**((8-NSH)*6)
        MMN2=MSK2-1
        DO 35 I=1,2
        IW(I)=M(KNT+I-1).AND.MMN2
        IW(I)=IW(I)*MSK1
        IC=M(KNT+1)/MSK2
35  IW(I)=IW(I).OR.(IC.AND.MMN1)
        GO TO 40
37  IW(1)=M(KNT)
        IW(2)=M(KNT+1)
40  ITEST=IW(2).AND.7700000000B
        IF(ITEST.EQ.8H00000000)GO TO 198
        ITEST=IW(1).AND.770000000000000000B
        IF(ITEST.EQ.8H00000000)IW(1)=IW(1).AND.777777777777777B
        DECODE(16,4,IW)ID,NRUN,NANT,IC,IData
        IF(IOHERR(0))198,43

```



```

43 IF(IC.NE.158)GO TO 198
   IOA(KOA)=IW(1)
   IOA(KOA+1)=IW(2)
   KOA=KOA+2
   LINE=LINE+1
   IF(NRUN.EQ.NRUNO.AND.LINE.LE.67)GO TO 53
   PRINT 5,NREC,(MSG(I),I=1,3),ID,NRUN
   LINE=1
53 NANT=NANT+1
   K(NANT)=K(NANT)+1
   IHZ(NANT)=IHZ(NANT)+IDATA
   IHZSQ(NANT)=IHZSQ(NANT)+IDATA*IDATA
   NRUNO=NRUN
   GO TO (55,56,57)NANT
55 PRINT 6,IDATA
   GO TO 205
56 PRINT 7,IDATA
   GO TO 205
57 PRINT 8,IDATA
   GO TO 205
59 STOP
   END

```

```

PROGRAM AVMODES
DIMENSION M(5000),TIME(5,3),TIMESQ(5,3),KNT(5,3),MODE(5,2)
DATA(((MODE(I,J),I=1,5),J=1,2)=1100,2900,4200,5100,6500,1655,3900,
15100,6060,7800)
CAL IOHCHECK
NF=0
READ 1,NFSKIP
1 FCXMAI(15)
IF(EOF,60)10,102
102 DO 105 I=1,NFSKIP
105 CALL SKIPFILE(1)
10 BUFFERIN(1,1)(M,M(10))
NREC=0
NF=NF+1
11 IF(UNIT,1)11,15,99,13
13 PRINT 2,NREC,NF
2 FORMAT(*OPARITY ERROR IN RECORD *,I3,* OF FILE *,I3)
15 IDATE=M(1).AND.777777777B
IDATE=IDATE*10000000B
M(2)=M(2)/10000000000B
IDATE=IDATE.OR.(M(2).AND.777777B)
PRINT 6
6 FORMAT(1H1/)
16 DO 17 J=1,3
DO 17 I=1,5
KNT(I,J)=0
TIME(I,J)=0.
17 TIMESQ(I,J)=0.
BUFFERIN(1,1)(M,M(5000))
NREC=NREC+1
21 IF(UNIT,1)21,25,10,23
23 PRINT 2,NREC,NF
25 L=LENGTHF(1)
DO 40 J=1,L,2
3 FORMAT(2I2,I1,R1,I6,4X)
DECODE(16,3,M(J))IDC,IRC,IA,ICOL,ITIME
IF(IOHERR(0))40,26
26 IF(ICOL.NE.15B)GO TO 40
IF(IA.LT.0.OR.IA.GT.2)GO TO 40
IAC=IA+1
DO 30 I=1,5
IF(ITIME.GT.MODE(I,2))GO TO 30
IF(ITIME.GE.MODE(I,1))GO TO 32
GO TO 40
30 CONTINUE
GO TO 40
32 KNT(I,IAC)=KNT(I,IAC)+1
PN=ITIME
PN=PN/100.
TIME(I,IAC)=TIME(I,IAC)+PN
TIMESQ(I,IAC)=TIMESQ(I,IAC)+PN**2
40 CONTINUE
50 DO 55 J=1,3
IA=J-1

```

```

DO 55 I=1,5
PN=KNT(I,J)
IF(KNT(I,J).EQ.0)GO TO 55
IF(KNT(I,J).GT.1)GO TO 52
TIMESQ(I,J)=0.
GO TO 53
52 TIMESQ(I,J)=SQRT(ABS(PN*TIMESQ(I,J)-TIME(I,J)**2)/(PN*(PN-1.)))
53 TIME(I,J)=TIME(I,J)/PN
PUNCH 4,IDATE,IDC,IRC,IA,I,TIME(I,J),TIMESQ(I,J),KNT(I,J)
4 FORMAT(A8,2I3,2I2,F7.3,E12.4,I5)
PRINT 5,IDATE,IDC,IRC,IA,I,TIME(I,J),TIMESQ(I,J),KNT(I,J)
5 FORMAT(1X,A8,2I3,2I2,F7.3,E12.4,I5)
55 CONTINUE
GO TO 16
99 STOP
END

```

## APPENDIX E

### UNIFORM DENSITY HYDROGEN DATA ANALYSIS AND ACCURACY STATEMENTS

The data analysis for uniform density hydrogen is based on 41 observations at tank pressures above 400 psi; the temperatures at the top of the tank were slightly lower than the bottom temperatures (the heater was at the bottom) indicating almost uniform temperature with a condition for convective mixing due to gravity; experimentally, these conditions were necessary to achieve a near uniform density within the experimental vessel.

The data for the  $TM_{011}$  mode, as plotted in Figure 6 of the main text, is expected to follow equation (6), i. e. ,

$$M = \frac{V}{A} \left( \frac{f_{01}^2 - f_1^2}{f_{01}^2 + 2f_1^2} \right) \quad (E-1)$$

Using equation (2), the expression for  $f_1$  in the time domain

$$f_1 = f_0 + r(t_1 - t_0) \quad (E-2)$$

it follows after a little algebra

$$t_1 - t_0 = \frac{f_{01}}{r} \sqrt{\frac{1 - \frac{AM}{V}}{1 + \frac{2AM}{V}}} - \frac{f_0}{r} \quad (E-3)$$

or equivalently

$$\Delta t_1 = \Delta t_0 + \frac{f_{01}}{r} \left( \sqrt{\frac{1 - \frac{AM}{V}}{1 + \frac{2AM}{V}}} - 1 \right) \quad (E-4)$$

where

$$\Delta t_1 = t_1 - t_0 \quad (E-5)$$

$$\Delta t_0 = \frac{f_{01} - f_0}{r}$$

( $\Delta t_0$  is just the time interval at  $M = 0$ ). The gauging function is then between  $\Delta t_1$  and  $M$ ; these quantities being related to three independent parameters  $\Delta t_0$ ,  $\frac{f_{01}}{r}$ ,  $\frac{A}{V}$  through equation (E-4). These three parameters may be determined by fitting the data to (E-4) or, alternately, may be determined by separate physical measurement. For example, the sweep rate,  $r$ , can be determined from  $\Delta t_0$ ,  $f_0$  and  $f_{01}$  by

$$r = \frac{f_{01} - f_0}{\Delta t_0} \quad (E-6)$$

The measured values of these quantities were  $f_{01} = 581.9$  MHz,  $f_0 = 411$  MHz and  $\Delta t_0 = 16.20$  milliseconds giving  $r = 10.54$  MHz/msec.  $V$  can be calculated from the inside radius of the tank  $R = 8.906$  inches (22.62 cm) giving  $V = 4.85 \times 10^4$  cm<sup>3</sup>.  $A$  can be obtained from other experimental data; ref. 4 of the main text gives  $A = 1.006$  cm<sup>3</sup>/gm. Table E-I gives the comparison between these values and the values obtained by a nonlinear least squares fit of the data to equation (E-4).\*

\* C. Daniel, F. S. Wood. Fitting Equations to Data, Wiley Interscience, N. Y. (1971) p. 320.

TABLE E-I

	$\Delta t_0$	$\frac{f_{01}}{r}$	$\frac{A}{V}$
Measured value	16.20 msec	55.15 msec	$9.42 \times 10^{-3} \text{ lb}^{-1}$
Fitted value	16.19	52.88	$9.44 \times 10^{-3}$
95% confidence limit (lower)	16.20	39.20	$6.81 \times 10^{-3}$
95% confidence limit (upper)	16.20	66.50	$12.21 \times 10^{-3}$

Although the confidence limits for the fitted values are relatively large, the actual fitted values are quite consistent with the independently determined values.

The semi-empirical gauge equation may then be given by

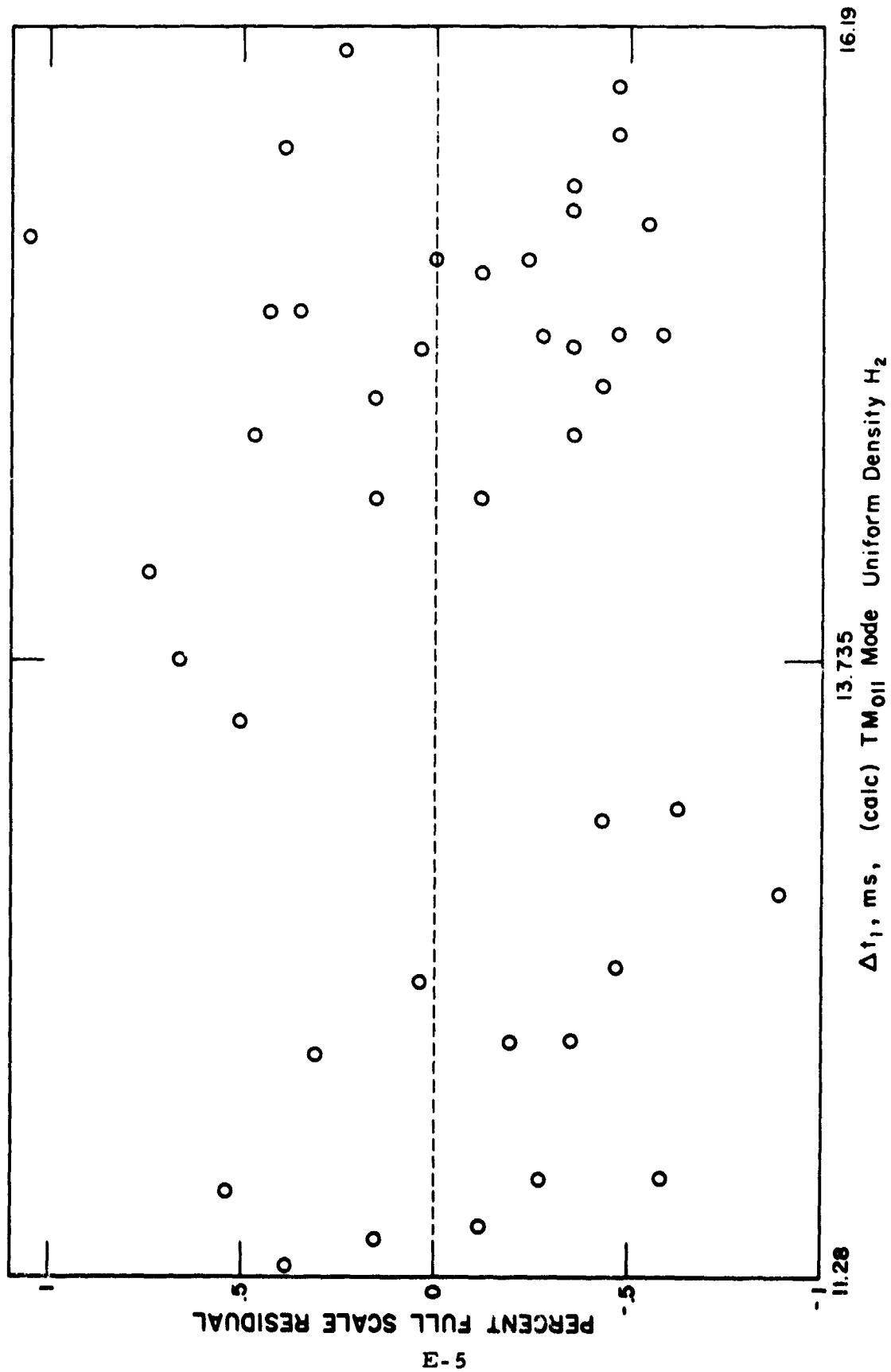
$$\Delta t_1 = 16.19 + 52.88 \left( \sqrt{\frac{1 - 9.44 \times 10^{-3} M}{1 + 2 \times 9.44 \times 10^{-3} M}} - 1 \right) \quad (\text{E-7})$$

where  $\Delta t_1$  is measured in milliseconds and  $M$  in pounds; it remains to determine the accuracy of the data with respect to this equation.

Table E-II gives the observed value of  $\Delta t_1$  and the value of  $\Delta t_1$  calculated from (E-7) for each of the observed mass values. The residuals are plotted in Figure E-1 as the Percent Full Scale Residual

TABLE E-II

<u>Mass (lbs)</u>	<u><math>\Delta t_1</math> (OBS)</u>	<u><math>\Delta t_1</math> (CALC)</u>	<u>Residual</u>	<u>% Full Scale Residual</u>
0.00	16.200	16.186	0.014	0.28%
0.20	16.018	16.037	-0.019	-0.38
0.44	15.839	15.859	-0.020	-0.41
0.52	15.820	15.799	0.021	0.43
0.73	15.631	15.644	-0.013	-0.26
0.84	15.550	15.564	-0.014	-0.29
0.94	15.460	15.483	-0.023	-0.47
1.00	15.500	15.446	0.054	1.10
1.15	15.340	15.337	0.003	0.06
1.15	15.329	15.337	-0.008	-0.16
1.16	15.326	15.330	-0.004	-0.08
1.42	15.163	15.141	0.022	0.45
1.42	15.160	15.141	0.019	0.38
1.54	15.034	15.054	-0.020	-0.41
1.54	15.029	15.054	-0.025	-0.51
1.56	15.029	15.039	-0.010	-0.20
1.57	15.019	15.032	-0.013	-0.26
1.58	15.030	15.025	0.005	0.10
1.81	14.842	14.859	-0.017	-0.35
2.06	14.674	14.687	-0.013	-0.26
1.85	14.841	14.831	0.010	0.20
2.11	14.670	14.645	0.025	0.51
2.43	14.415	14.417	-0.002	-0.04
2.45	14.413	14.403	0.010	0.20
2.82	14.180	14.142	0.038	0.78
3.33	13.820	13.786	0.034	0.69
3.70	13.557	13.530	0.027	0.55
4.23	13.141	13.167	-0.026	-0.53
4.25	13.136	13.154	-0.018	-0.38
4.71	12.815	12.856	-0.041	-0.83
5.19	12.501	12.520	-0.019	-0.39
5.20	12.518	12.514	0.004	0.08
5.62	12.221	12.235	-0.014	-0.28
5.63	12.222	12.228	-0.006	-0.12
5.68	12.212	12.195	0.017	0.35
6.43	11.679	11.703	-0.024	-0.49
6.44	11.686	11.697	-0.011	-0.22
6.51	11.680	11.651	0.029	0.59
6.77	11.480	11.483	-0.003	-0.06
6.79	11.480	11.470	0.010	0.20
7.08	11.304	11.283	0.021	0.43





vs.  $\Delta t_1$  (calc); this graph shows a fairly random scatter of the data between -0.83% and 1.10% full scale.

The cumulative distribution of the residuals is plotted in Figure E-2 vs. the expected intervals in a normal distribution. The S-shaped trend of the data indicates that the frequency distribution is slightly flatter than the normal bell-shaped error function. This is probably due to the fact that the  $\Delta t_1$  are really only measured to four significant figures, the most significant digit corresponding to 0.2%; this would cause a broadening of the frequency distribution on the order of  $\pm 0.2\%$ . The slope of the straight line through the tails of the curve gives an estimate of the standard deviation  $\sigma = 0.4\%$ . This gives a  $3\sigma$  deviation (the 99.9% confidence interval) of 1.20%.

#### Operational Readout

As indicated in the main text, a quadratic fit will be easier to work with operationally and should give a sufficiently accurate gauging function. The quadratic equation

$$\Delta t_1 = 16.20 + AM + BM^2 \quad (E-8)$$

was fitted to the data in Table II by solving (E-8) simultaneously for  $M = 6.77$ ,  $\Delta t_1 = 11.48$  and  $M = 2.43$ ,  $\Delta t_1 = 14.415$ . The result is  $A = -0.7554$  and  $B = 8.61 \times 10^{-3}$ . The residuals for this fit range between -0.63 and 0.93 percent full scale, a slightly better fit than the theoretical curve. This analysis is tabulated in Table III.

\* Chernov, H. and Lieberman, G. S., J. Amer. Statist. Assn. 49, 778-85 (1954).

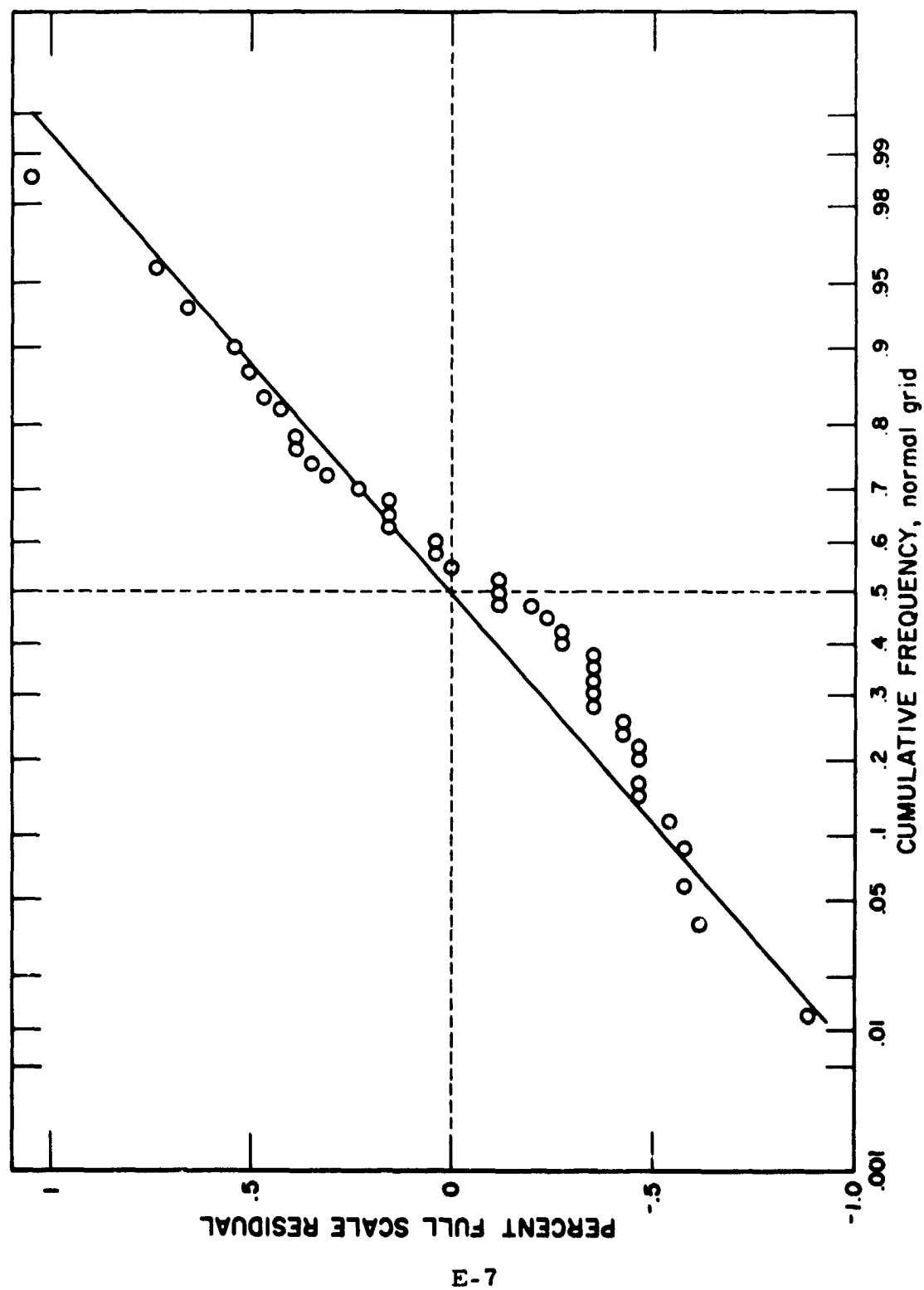


Figure E2. Cumulative Distribution of Residuals

TABLE E-III

<u>Mass (lbs)</u>	<u><math>\Delta t_1</math> (OBS)</u>	<u><math>\Delta t_1</math> (CALC)</u>	<u>% Full Scale Residual</u>
0.00	16.20	16.2	0.00%
0.2	16.018	16.049	0.63
0.44	15.839	15.869	0.62
0.52	15.82	15.809	0.21
0.73	15.631	15.653	-0.45
0.84	15.55	15.571	-0.44
0.94	15.46	15.497	-0.77
1.00	15.50	15.453	0.93
1.15	15.340	15.342	-0.05
1.15	15.329	15.342	-0.30
1.16	15.326	15.335	-0.19
1.42	15.163	15.144	0.37
1.42	15.160	15.144	0.31
1.54	15.034	15.057	-0.47
1.54	15.029	15.057	-0.57
1.56	15.029	15.042	-0.28
1.57	15.019	15.035	-0.33
1.58	15.03	15.027	0.04
1.82	14.842	14.853	-0.24
2.06	14.674	14.680	-0.13
2.11	14.67	14.644	0.52
2.43	14.415	14.415	0.01
2.45	14.413	14.400	0.25
2.82	14.180	14.138	0.85
3.33	13.820	13.780	0.82
3.7	13.557	13.522	0.70
4.23	13.141	13.158	-0.36
4.25	13.136	13.145	-0.19
5.19	12.501	12.511	-0.21
4.71	12.815	12.833	-0.37
5.2	12.518	12.504	0.27
5.62	12.221	12.226	-0.12
5.63	12.222	12.220	0.04
5.68	12.212	12.187	0.51
6.43	11.679	11.698	-0.40
6.44	11.688	11.692	-0.09
6.51	11.68	11.647	0.67
6.77	11.480	11.480	-0.01
6.79	11.480	11.467	0.25
7.08	11.304	11.283	0.42

## APPENDIX F

### TOTAL MASS GAUGING IN A SPHERICAL RESONANT CAVITY

#### Introduction

When a closed metal container is excited by an RF antenna probe inserted through a hole in the container, theoretically there are an infinite number of excitation frequencies for which the container is strongly coupled to the antenna; this means that energy can flow more freely between the antenna and the container at these resonant frequencies. The resonant frequencies correspond to standing wave patterns in the cavity which are called resonant modes. The wave pattern of the mode which occurs at the lowest possible resonant frequency is called the fundamental mode. This mode and the modes of the next few higher frequencies are called lower order modes.

When the cavity is uniformly filled with a fluid, the resonant frequency changes because the velocity of propagation of the resonant standing wave,  $c = 1/\sqrt{\mu\epsilon}$ , depends on the dielectric constant,  $\epsilon$ , and the magnetic permeability,  $\mu$ , of the fluid. For example, in a spherical resonant cavity uniformly filled, the resonant frequencies,  $f_{np}$ , are given by

$$f_{np} = \frac{u_{np}}{2\pi b\sqrt{\mu\epsilon}} \quad (F-1)$$

where  $b$  is the radius of the sphere, and  $n$  and  $p$  are subscripts which label the different modes (these will be explained in detail). The  $u_{np}$  are eigenvalues of the modes and are obtained in the process of finding solutions to Maxwells equations. The resonant frequencies can then be related to total mass by using the Clausius-Mossotti relation

$$P\rho = \frac{\epsilon - 1}{\epsilon + 2} \quad (F-2)$$

where  $P$  is the polarizability of the fluid which is a slowly varying function of the fluid density,  $\rho$ .

If the cavity is uniformly filled with a liquid the eigenvalues,  $u_{np}$ , are just numbers independent of the fluid within the cavity and there is therefore a simple relationship between resonant frequency and total mass. However, if the container is only partially filled with liquid, the rest of the cavity being a vacuum or a gas, then the values of  $u_{np}$  will depend on  $\mu$  and  $\epsilon$  of the liquid, the  $\mu_0$  and  $\epsilon_0$  of the gas, and the geometry which the liquid takes within the cavity; the resonant frequency, then, is no longer an unambiguous function of mass but depends on the liquid geometry as well. This is because the standing wave patterns are distorted because of the boundary conditions at the liquid-gas interface. However, the resonant frequency of each partially filled mode does lie between the completely empty and completely full values

$$\frac{u_{np}}{2\pi b\sqrt{\mu\epsilon}} \cdot f_{np} < \frac{u_{np}}{2\pi b\sqrt{\mu_0\epsilon_0}} \quad (F-3)$$

and varies continuously between these values as the cavity is filled. This suggests that the resonant frequency at least approximately indicates total mass independent of geometry.

The purpose of this note is to investigate the geometry effects for a spherical cavity with spherical symmetry of the liquid gas interface. This geometry is similar to a "zero-g" formation with the liquid clinging to the walls and a gas bubble in the middle of the cavity. The reason for choosing this geometry is that it is one of the few examples of a partially filled cavity for which the Maxwell Equations can be solved in closed form. Even though this geometry is particularly simple, it does give a reasonable

indication of the uncertainty which may be involved when the geometry is not known. Numerical examples are calculated for cases in which the liquid is hydrogen or nitrogen.

From a practical point of view, the spherical cavity is an ideal container geometry for this method of mass gauging. The reason for this is that the spherical symmetry of the cavity wall creates a degeneracy in the modes. That is, there are a number of standing wave patterns which have the same resonant frequency. This results in the fact that the distinct resonant frequencies of the lower order modes are widely separated and minimizes the effect of mode crossing in a partially filled cavity. Mode crossing occurs when, for a particular liquid geometry, the resonant frequency of a higher mode falls below that of a lower mode. For example, if the liquid is nitrogen, mode crossing between the first two modes is impossible and for the next few higher modes is quite unlikely; this is established from the table of eigenvalues, Table 1 on page F-12 and the inequalities expressed in (F-3).

The relative independence of the lower order modes suggests that they can each be monitored independently. Since each mode has its own geometry in the standing wave pattern, it seems reasonable that the modes themselves may be used to at least partially determine the fluid geometry. (Mathematically the problem reduces to this: Given some of the eigenvalues of a boundary value problem, how closely can the eigenfunctions be approximated.) In fact, it will be shown that for the spherical symmetry considered in this analysis, that for a liquid of unknown density, both the location of the liquid-gas interface and the density (hence the total mass) can be determined uniquely if and only if five modes are monitored simultaneously. The reason for this is that each mode determines exactly one independent relation between the resonant frequency

of that mode and the five unknown parameters  $\epsilon$ ,  $\mu$ ,  $\epsilon_0$ ,  $\mu_0$ , and  $a$ , where  $r = a$  is the liquid-gas interface. For most applications it is sufficient to assume that  $\epsilon_0 \approx \mu_0 \approx \mu \approx 1$ , leaving only two unknowns, namely  $\epsilon$  and  $a$ . In this case, two modes will uniquely determine the total mass.

#### Solutions for Maxwells Equations in Spherical Coordinates

When the cavity is resonating at an angular frequency,  $\omega$ , the time phase of the electromagnetic field is the same at all points within the cavity. Hence, for a loss free cavity the electric and magnetic fields can be written as the real parts of  $Ee^{i\omega t}$  and  $He^{i\omega t}$ , respectively, where  $E$  and  $H$  are vectors which depend only on the spacial coordinates. The source free Maxwell Equations can then be written

$$\begin{aligned}\text{curl } E &= -i\omega\mu H \\ \text{curl } H &= i\omega\epsilon E \\ \text{div } \epsilon E &= 0 \\ \text{div } \mu H &= 0.\end{aligned}\tag{F-4}$$

It should be emphasized at this point that only two assumptions have been made, the cavity is loss free and it is source free; in practice these are usually very good assumptions for calculating resonant frequencies. A third assumption which we will now make, may have to be justified more carefully in any given situation: we assume that there are two regions within the cavity, each of which have uniform density. The technical advantage of this assumption is that derivatives of  $\mu$  and  $\epsilon$  are not involved; the equation (F-4) can be solved in each region where  $\mu$  and  $\epsilon$  are constant and the boundary conditions are then modified to include the liquid-gas interface. The boundary conditions can be written

$$\{\epsilon E \cdot n, \mu H \cdot n, E_{\text{tn}} \text{ and } H_{\text{tn}} \text{ continuous at each boundary point}\} \tag{F-5}$$

where  $n$  is the unit normal vector to the surface at that point. Since

$\text{div } E = 0$  and  $\text{div } H = 0$ , both  $E$  and  $H$  can be expressed in terms of vector potentials  $G$  and  $F$ ,

$$\begin{aligned} E &= \text{curl } F \\ H &= \text{curl } G \end{aligned} \quad (\text{F-6})$$

where the Maxwell Equations impose consistency conditions between  $G$  and  $F$ . Two independent solutions may be obtained by choosing a coordinate direction, say  $\hat{r}$ , the unit vector in the radial direction and finding fields which are perpendicular to  $\hat{r}$ . If  $E$  is perpendicular to  $\hat{r}$  we say we have a TE (transverse electric) mode. This situation may be assured if  $F$  is chosen to be

$$F = f\hat{r} \quad (\text{F-7})$$

where  $f$  is a scalar function of the spatial coordinates. In this case we have from (F-4) and (F-7)

$$\begin{aligned} E &= \text{curl } f\hat{r} \\ H &= -\frac{1}{i\omega\mu} \text{curl curl } f\hat{r}. \end{aligned} \quad (\text{F-8})$$

If  $H$  is perpendicular to  $r$  we say we have a TM (transverse magnetic) mode. This situation may be assured if  $G$  is chosen to be

$$G = g\hat{r} \quad (\text{F-9})$$

where  $g$  is a scalar function of the spacial coordinates. In this case we have from (F-4) and (F-9)

$$\begin{aligned} E &= \frac{1}{i\omega\epsilon} \text{curl curl } g\hat{r} \\ H &= \text{curl } g\hat{r}. \end{aligned} \quad (\text{F-10})$$

The general solution for  $E$  and  $H$  may be obtained by a superposition of (F-8) and (F-10)



$$\begin{aligned}
E &= \text{curl } f\hat{r} + \frac{1}{i\omega\epsilon} \text{curl curl } g\hat{r} \\
H &= \text{curl } g\hat{r} - \frac{1}{i\omega\mu} \text{curl curl } f\hat{r} .
\end{aligned}
\tag{F-11}$$

To find equations which  $f$  and  $g$  satisfy, we consider the TE and TM modes separately. For the TM mode (F-10) and (F-4) imply that

$$\begin{aligned}
&\text{curl } E = -i\omega\mu \text{curl } g\hat{r} \\
&\text{or} \\
&\text{curl } (E + i\omega\mu g\hat{r}) = 0 .
\end{aligned}
\tag{F-12}$$

This last relation is satisfied only if

$$E + i\omega\mu g\hat{r} = \text{grad } \varphi \tag{F-13}$$

for some scalar function  $\varphi$ . Substituting (F-13) into the second of equation (F-4) we have

$$\text{curl curl } g\hat{r} = \omega^2\mu\epsilon g\hat{r} + i\omega\epsilon \text{grad } \varphi. \tag{F-14}$$

Using the vector equation

$$\text{curl curl } g\hat{r} = \nabla^2 g\hat{r} - \text{grad } (\nabla \cdot g\hat{r}) \tag{F-15}$$

and

$$\omega^2\mu\epsilon = k^2$$

we find that  $g$  satisfies the following equations

$$\begin{aligned}
(\nabla^2 + k^2) g\hat{r} &= 0 \\
\nabla \cdot g\hat{r} &= -i\omega\epsilon\varphi .
\end{aligned}
\tag{F-16}$$

A similar argument for the TE mode shows that  $f$  satisfies the following equation

$$\begin{aligned}
(\nabla^2 + k^2) f\hat{r} &= 0 \\
\nabla \cdot f\hat{r} &= i\omega\mu\nabla\psi .
\end{aligned}
\tag{F-17}$$

Equations (A-16) and (A-17) are equivalent to the scalar Helmholtz equations for  $g/r$  and  $f/r$  with standard solutions given by\*

$$B_n(kr) L_n^m(\theta, \varphi), \quad (F-18)$$

where the  $L_n^m(\theta, \varphi)$  are spherical harmonics and the  $B_n(kr)$  satisfy the differential equation

$$\left[ \frac{d^2}{dr^2} + k^2 - \frac{n(n+1)}{r^2} \right] B_n(kr) = 0. \quad (F-19)$$

The general solution of equation (F-19) can be given as a linear combination of  $j_n(kr)$  and  $y_n(kr)$  which are the Spherical Bessel Functions of order  $n$  of the first and second kind respectively.\*\*

$$B_n(kr) = C_n k r j_n(kr) + D_n k r y_n(kr) \quad (F-20)$$

where  $C_n$  and  $D_n$  are constants. The general solutions for  $f$  and  $g$  may be written as an infinite series

$$\begin{aligned} f &= \sum_{n,m} \left( C'_{nm} k r j_n(kr) + D'_{nm} k r y_n(kr) \right) L_n^m(\theta, \varphi) \\ g &= \sum_{n,m} \left( C_{nm} k r j_n(kr) + D_{nm} k r y_n(kr) \right) L_n^m(\theta, \varphi). \end{aligned} \quad (F-21)$$

The constants  $C_{nm}$ ,  $D_{nm}$ ,  $C'_{nm}$ , and  $D'_{nm}$  may be evaluated by substituting (F-21) into (F-11) and applying the boundary conditions (F-5). Equation (F-21) can be viewed as an infinite superposition of modes.

\* See R. F. Harrington, Time Harmonic Electromagnetic Fields, McGraw Hill (1961).

\*\* See M. Abrahamowitz and I. A. Stegun, NBS Handbook of Mathematical Functions, p. 437.

### TM Modes Under Spherical Symmetry

The TM modes are obtained by setting  $f = 0$ . If the liquid has spherical symmetry the boundary conditions may be satisfied by using only one value each for  $n$  and  $m$  in equation (F-21) and thus the series for  $g$  contains at most two non-vanishing terms:

$$g = \left( C_{nm} kr j_n(kr) + D_{nm} kr y_n(kr) \right) L_n^m(\theta, \varphi). \quad (F-22)$$

Using equations (F-11) and (F-16) along with (F-22), the components of the electric and magnetic fields may be written as follows:

$$\begin{aligned} E_r &= \frac{-1}{i\omega\epsilon} \left( \frac{\partial^2}{\partial r^2} + k^2 \right) g = -\frac{n(n+1)}{i\omega\epsilon r^2} g \\ E_\theta &= \frac{-1}{i\omega\epsilon r} \frac{\partial^2}{\partial r \partial \theta} g \\ E_\varphi &= \frac{-1}{i\omega\epsilon r \sin \theta} \frac{\partial^2}{\partial r \partial \varphi} g \\ H_r &= 0 \\ H_\theta &= -\frac{1}{r \sin \theta} \frac{\partial g}{\partial \varphi} \\ H_\varphi &= \frac{1}{r} \frac{\partial g}{\partial \theta}. \end{aligned} \quad (F-25)$$

The boundary conditions are applied by letting the container walls exist at  $r = b$  and the liquid-gas interface at  $r = a \leq b$  (if  $a = 0$  the container is full and if  $a = b$  the container is empty.). The conditions  $\mu H \cdot n$  and  $H \times n$  continuous at  $r = a$  and  $r = b$  imply continuity of  $H_\theta$  and  $H_\varphi$  and hence that  $g$  is continuous at  $r = a$  and  $r = b$ . This is compatible with the continuity of  $\epsilon E \cdot n$ . The condition  $E \times n$  continuous implies that  $E_\theta$  and  $E_\varphi$  is continuous and hence that  $\frac{1}{\epsilon} \frac{\partial}{\partial r} g$  is continuous at  $r = a$  and  $r = b$ . In summary the boundary conditions are completely specified by

$$g \text{ continuous at } r = a \text{ and } r = b \quad (F-26)$$

$$\frac{1}{\epsilon} \frac{\partial}{\partial r} g \text{ continuous at } r = a \text{ and } r = b. \quad (F-27)$$

Since the  $L_n^m(\theta, \varphi)$  are independent both of radial position and fluid properties, the condition (F-26) is equivalent to

$$k_o a j_n(k_o a) = C_{nm} k a j_n(ka) + D_{nm} k a y_n(ka) \quad (F-28)$$

and

$$g(b) = C_{nm} k b j_n(kb) + D_{nm} k b y_n(kb) \quad (F-29)$$

where the coefficient of  $y_n(k_o a)$  in equation (F-22) is zero because  $g$  must be finite at  $r = 0$ . (Here,  $k_o = \omega/\epsilon_o \mu_o$  applies to the region in the gas and  $k = \omega/\epsilon \mu$  applies to the region in the liquid.) Likewise condition (F-27) is equivalent to

$$\frac{1}{\epsilon_o} \frac{\partial}{\partial a} [k_o a j_n(k_o a)] = \frac{1}{\epsilon} \frac{\partial}{\partial a} [D_{nm} k a j_n(ka) + D_{nm} k a y_n(ka)] \quad (F-30)$$

and

$$0 = \frac{1}{\epsilon} \frac{\partial}{\partial b} [C_{nm} k b j_n(kb) + D_{nm} k b y_n(kb)]. \quad (F-31)$$

Equations (F-28), (F-30), and (F-31) are three independent relations in the eight variables,  $C_{nm}$ ,  $D_{nm}$ ,  $a$ ,  $\omega$ ,  $\epsilon$ ,  $\mu$ ,  $\epsilon_o$ , and  $\mu_o$ . The inhomogeneous equations (F-28) and (F-30) can be solved uniquely for  $C_{nm}$  and  $D_{nm}$  and these values are substituted in equation (F-31) which then becomes a homogeneous relation in six variables  $a$ ,  $\omega$ ,  $\epsilon$ ,  $\mu$ ,  $\epsilon_o$ , and  $\mu_o$ . We will denote this relation by

$$F_n(\omega, a, \epsilon, \mu, \epsilon_o, \mu_o) = 0 \quad (F-32)$$

or sometimes more simply by  $F_n(\omega, \text{etc.}) = 0$ . For a given set of values for  $a$ ,  $\epsilon$ ,  $\mu$ ,  $\epsilon_o$ , and  $\mu_o$  (which is determined by conditions in the container), it can be shown that  $F_n$  plotted as a function of  $\omega$  is oscillatory

and hence there are an infinite number of solutions to equation (F-32). The solution for the  $p^{\text{th}}$  zero of equation (F-32) is called  $\omega_{np}$  and the field pattern obtained by substituting  $\omega_{np}$  and the values for  $C_{nm}$  and  $D_{nm}$  into equations (F-22) and (F-25) is called the  $TM_{mnp}$  mode where

$$\begin{aligned} n &= 1, 2, 3, \dots \\ p &= 1, 2, 3, \dots \\ m &= 0, \pm 1, \pm 2, \dots, \pm n. \end{aligned}$$

(The range on  $m$  comes from the properties of the spherical harmonics.) Since  $\omega_{np}$  is independent of  $m$ , we see that there are a number of modes corresponding to the same  $\omega_{np}$ . This number is called the degeneracy of  $\omega_{np}$ . For example, the fundamental frequency  $\omega_{11}$  corresponds to three modes,  $TM_{011}$ ,  $TM_{-111}$ , and  $TM_{111}$  and hence has degeneracy 3. Sometimes the first subscript is dropped and the three modes are collectively referred to as the  $TM_{11}$  mode (which is an abuse of the term "mode").

We now discuss the conditions under which the resonant frequencies  $\omega_{np}$  can determine the total mass. The total mass  $M$  is a function of three of the above variables,  $a$ ,  $\epsilon_0$ , and  $\epsilon$ . If the resonant frequencies,  $\omega_{np}$ , of the modes are known, then we have the following relations in the five variables  $a$ ,  $\epsilon_0$ ,  $\mu_0$ ,  $\epsilon$ , and  $\mu$

$$\begin{aligned} 0 &= F_1(\omega_{11}, \text{etc.}) = F_1(\omega_{12}, \text{etc.}) = \dots \\ &= F_2(\omega_{21}, \text{etc.}) = F_2(\omega_{22}, \text{etc.}) = \dots \\ &\quad \vdots \\ &= F_n(\omega_{np}, \text{etc.}) \quad (F-33) \end{aligned}$$

where each of the  $F_n(\omega_{np}, \text{etc.})$  is a relation determined by measuring the resonant frequency of a  $TM_{mnp}$  mode. Since there are five variables,

it is clear that at least five different modes are necessary to completely determine the total mass. From the properties of the Spherical Bessel Functions it can be shown that each of the above relations is also independent; therefore, five modes are also sufficient to determine the total mass. If further assumptions are made, fewer modes may be sufficient. For example, for most liquids  $\mu \approx \mu_0 \approx 1$  reduces the number of necessary modes to 3; if it is further assumed that  $\epsilon_0 \approx 1$ , then the number of necessary modes is two; finally if in addition  $\epsilon$  is known, then only one resonant frequency is necessary to determine the total mass.

Alternately, it may be that the interface,  $r = a$ , is known and  $\epsilon$  (hence the density) is unknown; if  $\epsilon_0 \approx \mu_0 \approx \mu \approx 1$ , then the density and hence the total mass may be determined by a single resonant frequency. As a limiting case of this situation, the case  $a = 0$  indicates a completely full cavity and the resonant frequencies are given by

$$\omega_{np} = \frac{u_{np}}{b\sqrt{\epsilon\mu}} \quad (F-34)$$

where  $u_{np}$  is the  $p^{\text{th}}$  zero of equation (F-32) considered as a function of the quantity  $kb$ . (The quantities  $u_{np}$  are also known as eigenvalues of the  $TM_{np}$  "mode".) The measured frequency  $f_{np}$  is given by  $f_{np} = \frac{\omega_{np}}{2\pi}$ . The calculated values for  $u_{np}$  in the case  $a = 0$  are listed in Table 1 in increasing order for the lowest ten modes. (Table 1 also includes results of a similar analysis for the TE modes.) The resonant frequencies  $f_{np}$  also plotted in Table 1 are for the specific case of a 48-cm diameter empty container.

We see from Table 1 that the resonant frequencies of the lower order modes are widely spaced. This is primarily due to the degeneracy and makes it feasible to simultaneously monitor several of the lower order modes.

Table F-1.

Modes	Eigenvalues	Degeneracy	Frequency (48 cm dia. Sphere)	
TM <sub>11</sub>	$u_{11} = 2.744$	3	$f_{11}$	0.543 GHz
TM <sub>21</sub>	$u_{21} = 3.870$	5	$f_{21}$	0.766
TE <sub>11</sub>	$u'_{11} = 4.493$	3	$f'_{11}$	0.889
TM <sub>31</sub>	$u_{31} = 4.973$	7	$f_{31}$	0.984
TE <sub>21</sub>	$u'_{21} = 5.763$	5	$f'_{21}$	1.140
TM <sub>41</sub>	$u_{41} = 6.062$	9	$f_{41}$	1.200
TM <sub>12</sub>	$u_{12} = 6.117$	3	$f_{12}$	1.210
TE <sub>31</sub>	$u'_{31} = 6.998$	7	$f'_{31}$	1.384
TM <sub>51</sub>	$u_{51} = 7.140$	11	$f_{51}$	1.413
TM <sub>22</sub>	$u_{22} = 7.443$	5	$f_{22}$	1.472

#### Example: Using Hydrogen and Nitrogen

Equation (F-32) was solved for the four lowest order modes using the FORTRAN program listed in Table 2.\* For given values of  $a$ ,  $\epsilon$ ,  $\mu$ ,  $\epsilon_0$ , and  $\mu_0$ , the program finds the zeros of  $F_n(\omega, \text{etc.})$  plotted as a function of  $kb$  where

$$kb = \omega \sqrt{\mu \epsilon} b.$$

The  $p^{\text{th}}$  zero is

$$u_{np} = \omega_{np} \sqrt{\mu \epsilon} b.$$

The computer plots the quantity  $\alpha u_{np}$  vs.  $\bar{\omega}$  which is essentially resonant frequency,  $f_{np}$ , vs. total mass  $M$ . Here,

$$\alpha = \sqrt{\frac{\epsilon_0 \mu_0}{\epsilon \mu}}.$$

\* Written by A. E. Hiester.

Table F-II.

```

PROGRAM PLOT3
DIMENSION IFILM(13),ITITLE(13),X(100),Y(100),AL(3),RHO(3)
DATA (IFILM=24,HART HIESTER, X3474
A(N,U,F,AL)=(1./AL*PJ(N,AL*F*U)*YPP(N,F*U)-AL**2*PPJ(N,AL*F*U)*YP(
1N,F*U))/(PJ(N,F*U)*YPP(N,F*U)-PPJ(N,F*U)*YP(N,F*U))
B(N,U,F,AL)=(AL**2*PJ(N,F*U)*PPJ(N,AL*F*U)-1./AL*PPJ(N,F*U)*PJ(N,A
1L*F*U))/(PJ(N,F*U)*YPP(N,F*U)-PPJ(N,F*U)*YP(N,F*U))
FUN(N,U,F,AL)=A(N,U,F,AL)*PJ(N,U)-B(N,U,F,AL)*YP(N,U)
1 FORMAT(3F10.0)
2 FORMAT(*U*,12.5H )
3 FORMAT(1H1,10X,2A8//9X,11HALPHA * UNP,10X,*RHOBAR*//)
4 FORMAT(9X,F9.5,10X,F10.7)
5 FORMAT(*OU NOT FOUND IN 10J ITERATIONS*//1X,6E22.8)
6 FORMAT(5H$1* U,12,1H )
P=1.
ITITLE(1)=8H RESONAN
ITITLE(2)=8HT FREQUE
ITITLE(3)=8HNCY VS M
ITITLE(4)=8HASS - H2
ITITLE(7)=8HR$9HOBAR
ITITLE(10)=8H $1
ITITLE(11)=8HA$9LPHA
ITITLE(5)=ITITLE(6)=ITITLE(8)=ITITLE(9)=ITITLE(13)=8H
READ 1,(AL(I),I=1,3)
READ 1,(RHO(I),I=1,3)
CALL GRAPH(1,1,3,IFILM,0,6)
DO 60 N=1,4
ID=P
ID=ID+10*N
ENCODE(8,2,IFILM)ID
ENCODE(8,6,ITITLE(12))ID
DO 55 I=1,3
GO TO (7,8,9),I
7 LTYP=8HTP SOLID
GO TO 95
8 LTYP=8HTP LIQ
GO TO 95
9 LTYP=8HNBP LIQ
95 PRINT 3,IFILM(1),LTYP
LINE=0
DO 50 J=1,99
RHOBAR=J
RHOBAR=RHOBAR/100.*RHO(I)
F=(1.-RHOBAR/RHO(I))**(1./3.)
UB=7.5
US=2.5
FUS=FUN(N,US,F,AL(I))
FUB=FUN(N,UB,F,AL(I))
IT=0
10 UM=(UB-US)/2.+US
IT=IT+1
IF(IT.LE.100)GO TO 15
PRINT 5,US,FUS,UM,FUM,UB,FUB
STOP

```



Table F-II. (Continued)

```

15 FUM=FUN(N,UM,F,AL(I))
   IF (ABS(FUM).LT..00001)GO TO 45
   IF (FUM.GT.0..AND.FUS.GT.0..OR.FUM.LT.0..AND.FUS.LT.0.)GO TO 20
   UB=UM
   FUB=FUM
   GO TO 10
20 US=UM
   FUS=FUM
   GO TO 10
45 X(J)=RHOBAR
   Y(J)=AL(I)*UM
   LINE=LINE+1
   IF (LINE.NE.51)GO TO 50
   LINE=0
   PRINT 3,IFILM(1),LTP
50 PRINT 4,Y(J),X(J)
   IF (I.NE.1)GO TO 53
   CALL LGRAPH(X,Y,99,ITITLE,IFILM)
   CALL CPGRAPH(X(99),Y(99),1,,,I+4)
   GO TO 55
53 CALL CLGRAPH(X,Y,99)
   CALL CPGRAPH(X(99),Y(99),1,,,I+4)
55 CONTINUE
   IFILM(1)=8H$9, $1TP
   IFILM(2)=8H SOLID'/
   IFILM(3)=8H$1+ TP L
   IFILM(4)=8H LIQUID'/$
   IFILM(5)=8H1* NBP L
   IFILM(6)=8H LIQUID
   CALL COMGRAPH(.75,.75,6,IFILM)
   CALL SKIPFRM
60 CONTINUE
   STOP
   END

FUNCTION SY(N,Z)
  Y1(Z)=-COS(Z)/Z
  Y2(Z)=-COS(Z)/Z**2-SIN(Z)/Z
  Y3(Z)=(-3./Z**3+1./Z)*COS(Z)-3./Z**2*SIN(Z)
  GO TO (10,20,30,40,50)N+1
10 SY=Y1(Z)
   RETURN
20 SY=Y2(Z)
   RETURN
30 SY=Y3(Z)
   RETURN
40 SY=5./Z*Y3(Z)-Y2(Z)
   RETURN
50 SY=7./Z*(5./Z*Y3(Z)-Y2(Z))-Y3(Z)
   RETURN
   END

FUNCTION SJ(N,Z)
  J1(Z)=SIN(Z)/Z

```

Table F-II. (Continued)

```

J2(Z)=SIN(Z)/Z**2-COS(Z)/Z
J3(Z)=(3./Z**3-1./Z)*SIN(Z)-3./Z**2*COS(Z)
GO TO (10,20,30,40,50)N+1
10 SJ=J1(Z)
   RETURN
20 SJ=J2(Z)
   RETURN
30 SJ=J3(Z)
   RETURN
40 SJ=5./Z*J3(Z)-J2(Z)
   RETURN
50 SJ=7./Z*(5./Z*J3(Z)-J2(Z))-J3(Z)
   RETURN
   END

FUNCTION PJ(N,Z)
  FN=N
  PJ=Z*SJ(N-1,Z)-FN*SJ(N,Z)
  RETURN
  END

FUNCTION YP(N,Z)
  FN=N
  YP=Z*SY(N-1,Z)-FN*SY(N,Z)
  RETURN
  END

FUNCTION PPJ(N,Z)
  FN=N
  PPJ=(FN*(FN+1.))/Z*SJ(N,Z)
  RETURN
  END

FUNCTION YPP(N,Z)
  FN=N
  YPP=(FN*(FN+1.))/Z*SY(N,Z)
  RETURN
  END

```

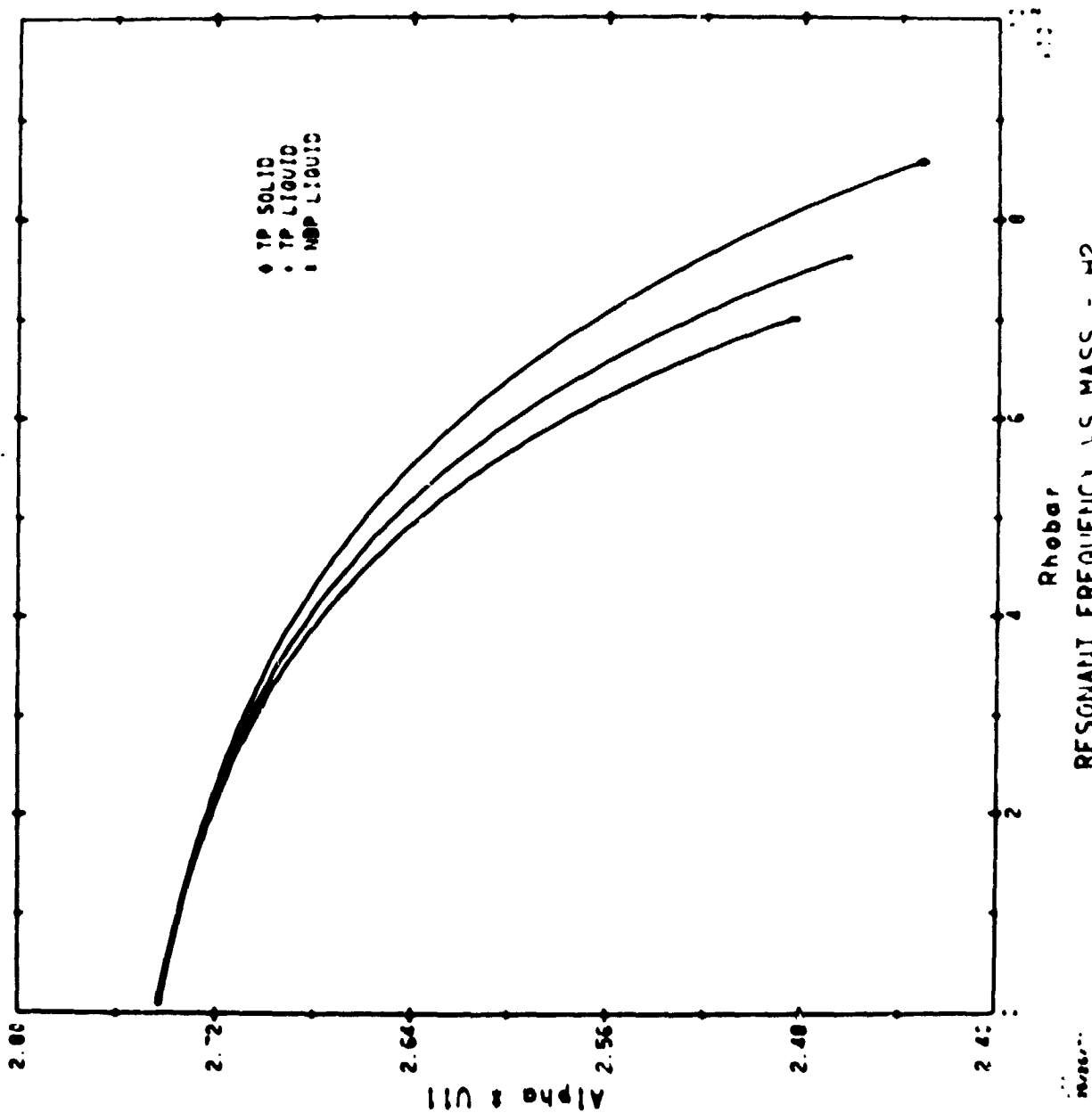
$$f_{np} = \frac{\alpha u_{np}}{2\pi b \sqrt{\epsilon_0 \mu_0}}$$

and

$$M = \delta V$$

where  $V$  is the volume of the tank in  $\text{cm}^3$ . The results may then be applied to spheres of any size and to any dielectric fluid.

We have assumed that  $\mu = \mu_0 = \epsilon_0 = 1$  and plotted the results for three different densities corresponding to solid hydrogen, triple point liquid, and normal boiling point liquid; this corresponds to about 22 percent range in density. The results for the first four modes are shown in figures A1, A2, A3, and A4. It is seen that the uncertainty in total mass is smaller for higher modes. Qualitatively this is because the field patterns are spread more uniformly throughout the cavity for the higher modes. For example, the uncertainty in mass vs.  $\alpha u_{41}$  (or  $f_{41}$ ) is less than 5 percent over most of the range. This is to be compared with a density change of 22 percent indicating that the resonant mode has a tendency to integrate over the mass of the liquid rather than the volume.



RESONANT FREQUENCY VS MASS - H2

Figure F1. Resonant Frequency vs. Mass,  $TM_{m11}$  Mode.

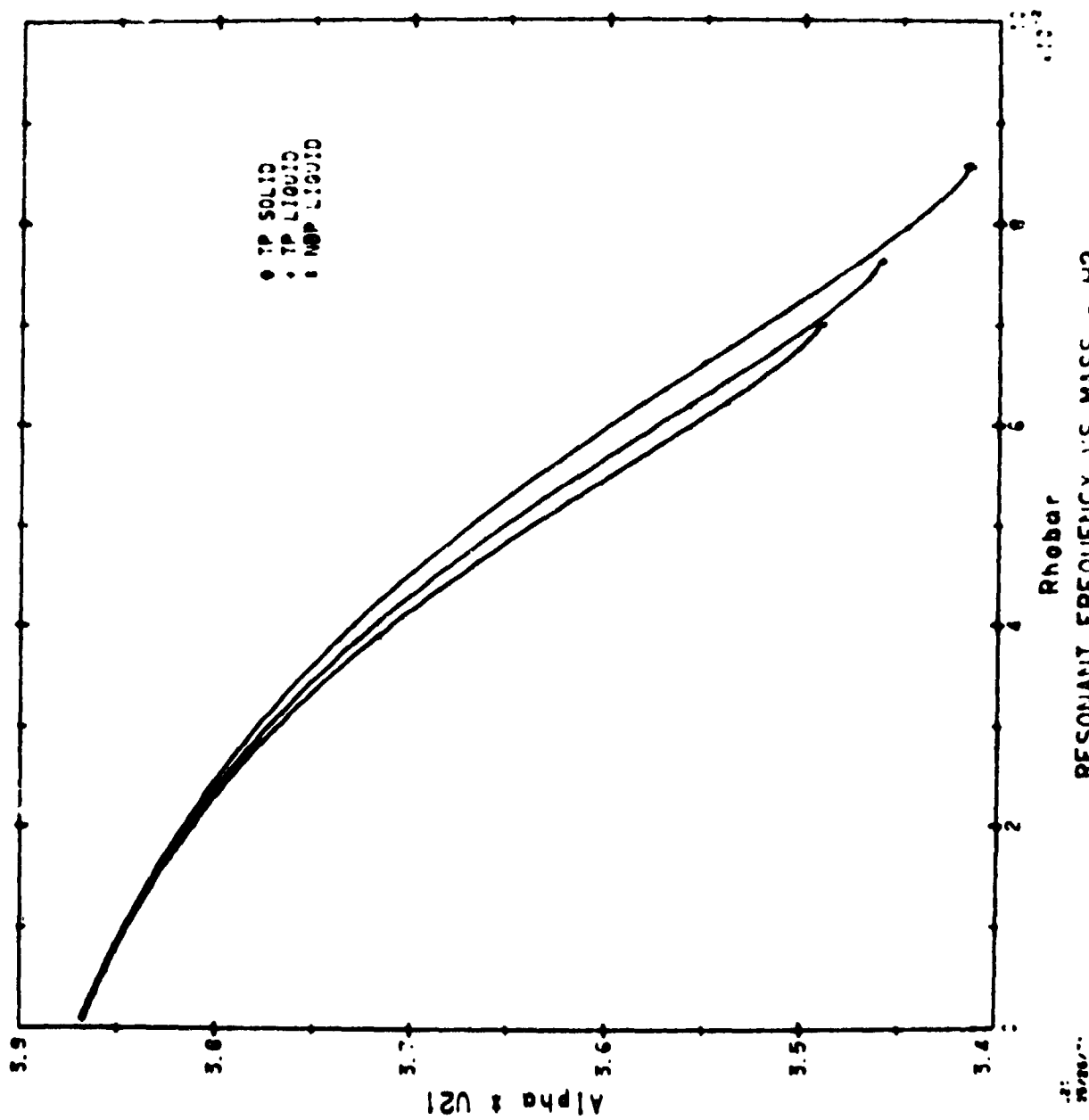
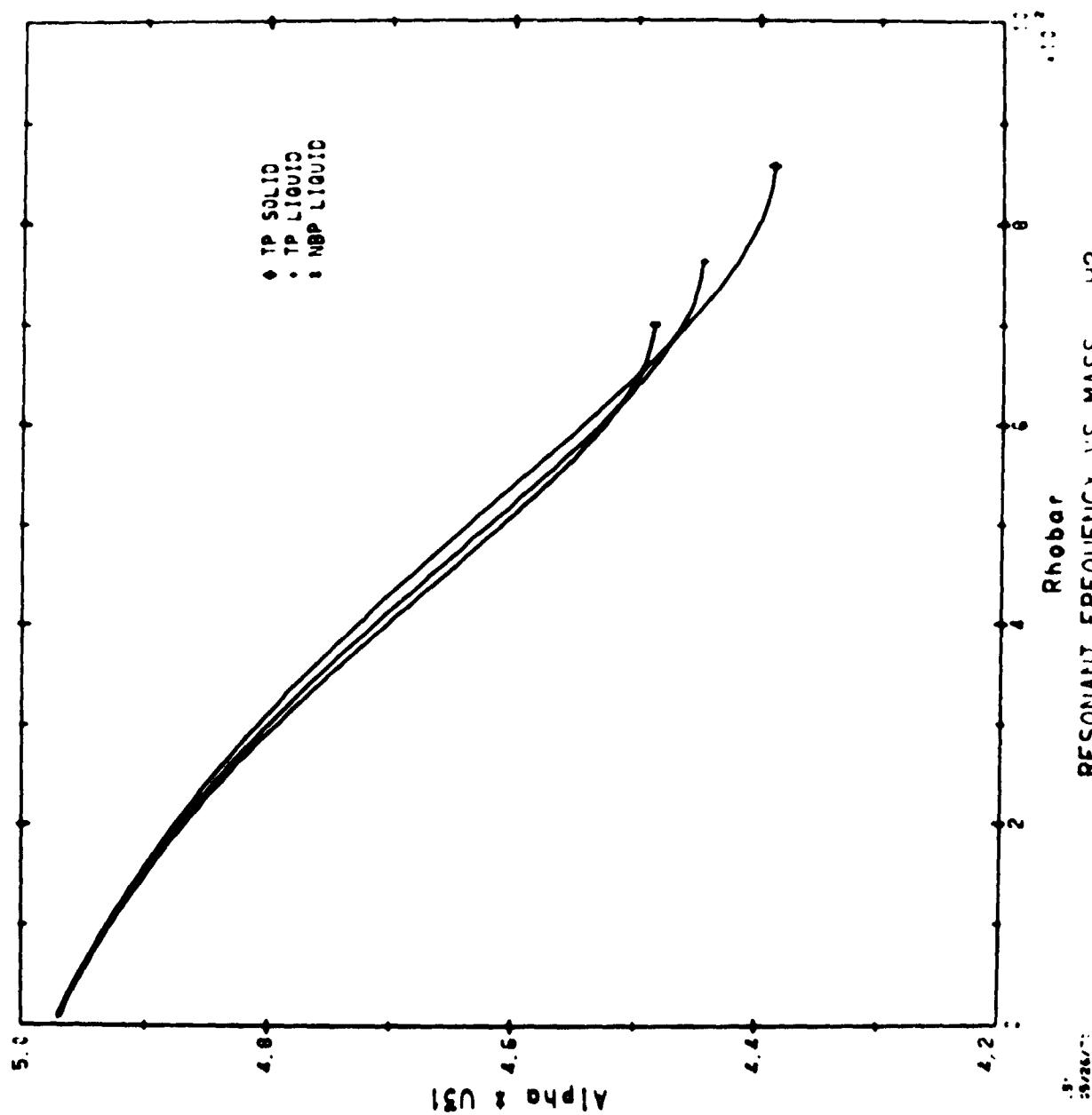


Figure F2. Resonant Frequency vs. Mass,  $TM_{m21}$  Mode.



RESONANT FREQUENCY VS MASS - H2

Figure F3. Resonant Frequency vs. Mass, TM<sub>m31</sub> Mode.

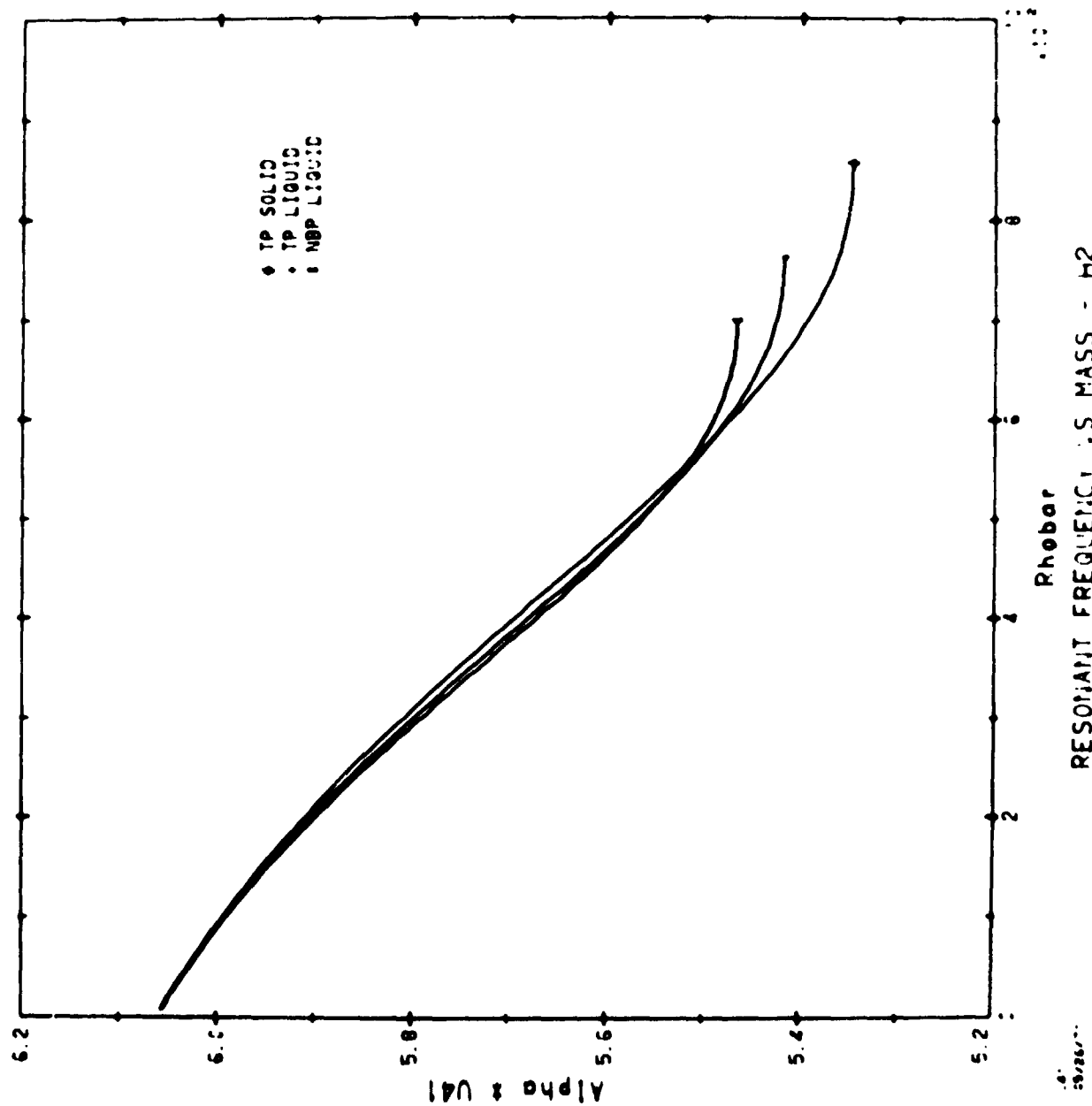


Figure F4. Resonant Frequency vs. Mass,  $TM_{m41}$  Mode.

## APPENDIX G

### APPROXIMATE METHODS FOR AN INHOMOGENEOUS DIELECTRIC

#### Introduction

The problem is to compute resonant frequencies of a microwave cavity containing an inhomogeneous but isotropic dielectric. We suppose the cavity wall is a perfect conductor, the dielectric dissipates no power, and the dielectric has uniform magnetic permeability. Mathematically, we are dealing with the boundary value problem posed by Maxwell's equations<sup>1</sup> in the absence of sources and with the electric vector everywhere normal to the wall of the cavity. For a Fourier component  $Ee^{j\omega t}$  of electric field with (angular) frequency  $\omega$ , the boundary value problem is

$$\begin{aligned}\nabla \times E &= -j\omega\mu H \\ \nabla \times H &= j\omega\epsilon E\end{aligned}\tag{1}$$

$E$  normal to boundary.

We will use the subscripts 0, 1 to denote quantities pertaining to the corresponding mode for a cavity containing a uniform dielectric of permittivity  $\epsilon_0$  or  $\epsilon_1$ , respectively. We assume that the permittivity  $\epsilon$  is piecewise continuous and satisfies  $\epsilon_0 \leq \epsilon \leq \epsilon_1$ . Also we assume that  $\epsilon_1 - \epsilon_0$  is small enough so that the set of modes for the permittivities  $\epsilon_0$ ,  $\epsilon$ ,  $\epsilon_1$  are at most slightly different from each other in shape and can be put in one-to-one correspondence. For convenience we will regard  $\epsilon_0$  as the permittivity of free space.

---

<sup>1</sup> We use the technique, notation, and units (MKSA) of Wolfgang K. H. Panofsky and Melba Phillips, Classical Electricity and Magnetism, Addison-Wesley, 1955.



The resonant frequency  $\omega_0$  of any mode in the empty cavity may be computed<sup>2</sup> from a standard solution of Maxwell's equations. The resonant frequency  $\omega$  of the corresponding mode when the cavity is partially filled with dielectric is lower, and we wish to estimate its value without further extensive calculations. We prefer estimating techniques which are insensitive to the spatial distribution of dielectric material in the cavity and which do not require further computation of electromagnetic field strengths.

The first method considered is adapted from a technique of approximation due to Rayleigh. Two approximations of this type are formed, and it is shown that one is always at least as large as the other. Then the larger is shown to be always at least as large as the true value of  $\omega$ . Finally, a refinement of the last method is described, known as the Rayleigh-Ritz method. This first group of methods gives upper bounds for the true resonance frequency. The lower Rayleigh estimate has not been proved to be a lower bound. But the difference between the upper and lower Rayleigh estimates is within 10 percent of the difference between the empty and full cavity resonant frequencies, in the case of liquid nitrogen in a sphere in a steady uniform gravitational field (See Figure G1). One should bear in mind that the upper Rayleigh estimate has been proved an upper bound only for the fundamental ( $TM_{011}$ ) mode.

The moment methods improve on the Rayleigh methods in two ways. First, they provide lower bounds as well as upper bounds; and second, they are easier to apply to higher modes. The first order moment method yields a lower bound for the fundamental mode which is comparable to the lower Rayleigh estimate, and an upper bound which is identical to the upper Rayleigh estimate. Higher moments have not yet been computed.

---

<sup>2</sup>See Appendix F, Table I.

The Green's function was investigated as a tool for obtaining lower bounds. The resonance frequencies of the cavity are eigenvalues of a certain linear operator  $L$ . The Green's function is used to compute a norm for  $L^{-1}$ . Preliminary results are inconclusive as to its value, and careful estimates must be made with an automatic computer.

### Rayleigh Methods

The Rayleigh method<sup>3</sup> of equating "potential" and "kinetic" energies in the perturbed field suggests the following heuristic procedure. We suppose that  $E = a E_0$  for some number  $a$  independent of position and compute the magnetic field from equations (1):

$$H = \frac{\nabla \times E}{-j\omega\mu} = \frac{a \nabla \times E_0}{-j\omega\mu} = \frac{a(-j\omega_0\mu H_0)}{-j\omega\mu} = a \frac{\omega_0}{\omega} H_0.$$

Then we eliminate  $a$  by equating the time-average electric and magnetic field energies<sup>4</sup>.

$$\int \frac{1}{2} \epsilon E^2 \approx \int \frac{1}{2} \mu H^2$$

$$a^2 \int \epsilon E_0^2 \approx a^2 \int \mu \frac{\omega_0^2}{\omega^2} H_0^2.$$

But  $\int \epsilon_0 E_0^2 = \int \mu H_0^2$ , so we conclude that  $\omega^2$  is approximately equal to

$$\omega^2 = \omega_0^2 \frac{\int E_0^2}{\int K E_0^2} \quad (2)$$

where  $K = \epsilon/\epsilon_0$  is the dielectric constant. Notice that the ratio  $\omega$  of  $H$  to  $H_0$  is also independent of position, as  $a$  is.

<sup>3</sup>G. Temple and W. G. Bickley, Rayleigh's Principle, Dover, 1956, pp. 1-24.

<sup>4</sup>All integrals are volume integrals over the region of the cavity, unless otherwise specified.

But if we suppose  $H = bH_0$  for some number  $b$  independent of position, equations (1) lead to

$$E = b \frac{\omega_0 \epsilon_0}{\omega \epsilon} E_0$$

and the ratio of  $E$  to  $E_0$  is now dependent on position because  $\epsilon$  is. A computation like that of the preceding paragraph shows that  $\omega^2$  is approximately equal to

$$\frac{\omega^2}{\omega_0^2} = \frac{\int \frac{1}{K} E_0^2}{\int E_0^2} \quad (3)$$

We show (Theorem 1) that  $\omega_1 < \underline{\omega} \leq \bar{\omega} \leq \omega_0$  for every mode and (Theorem 2) that  $\omega_1 \leq \omega \leq \bar{\omega}$  for the fundamental mode. The remainder of this Appendix describes ways to compute lower bounds for  $\omega$  in the fundamental mode.

The electric field  $E$  for any solution of (1) for a given  $\epsilon$  and  $\omega$  is an eigenfunction of the differential operator

$$L = \frac{1}{\mu \epsilon} \nabla \times \nabla \times$$

corresponding to the eigenvalue  $\omega^2$ . Let  $S$  be the spherical cavity including the inside region and also the boundary surface. Let  $\Delta$  be the set of all vector functions on  $S$  having continuous derivatives of all orders. Then  $\Delta$  is a pre-Hilbert space with respect to the inner product

$$(F, G) = \int F \cdot G,$$

and we will denote by  $\Theta$  the Hilbert space which is the  $(,)$  - completion of  $\Delta$ . It is easy to show that  $L$  maps  $\Delta$  into  $\Theta$ , and so  $L^{-1}$  is well-defined on  $L\Delta$ . But  $L^{-1}$  is bounded (Theorem 3) and  $L\Delta$  is dense so that  $L^{-1}$  has a unique continuous extension to  $\Theta$  which we also denote  $L^{-1}$ .

Now  $L$  is symmetric<sup>5</sup> in the sense that

$$(LF, G) = (F, LG)$$

for every  $F, G$  in  $\Delta$  and positive in the sense that  $(LF, F) > 0$  for every  $F$  in  $D$ . The same is true for  $L_1, L_0$ , and the inverses of all three operators. Moreover we have easily

$$(E_0, LE_0) \leq (E_0, L_0 E_0)$$

$$(E_1, L^{-1} E_1) \leq (E_1, L_1^{-1} E_1).$$

Now observe that

$$L_0 - \omega_0^2 = \frac{\epsilon_1}{\epsilon_0} \left( L_1 - \omega_0^2 \frac{\epsilon_0}{\epsilon_1} \right),$$

so that  $E_0, E_1$  belong to the same eigenspace of  $L_0$ . But we have agreed<sup>6</sup> that  $E_0, E, E_1$  shall correspond to exactly the same mode. This means that  $E_1$  is a scalar multiple of  $E_0$  and

$$\frac{(E_1, E_1)}{(E_1, L^{-1} E_1)} = \frac{(E_0, E_0)}{(E_0, L^{-1} E_0)}.$$

Theorem 1:  $\omega_1 \leq \underline{\omega} \leq \bar{\omega} \leq \omega_0.$

Proof:

$$\begin{aligned} \omega_1^2 &= \frac{(E_1, E_1)}{(E_1, \omega_1^{-2} E_1)} = \frac{(E_1, E_1)}{(E_1, L_1^{-1} E_1)} \\ &\leq \frac{(E_1, E_1)}{(E_1, L^{-1} E_1)} \cdot \frac{(E_0, E_0)}{(E_0, L^{-1} E_0)} = \frac{(E_0, E_0)}{(E_0, K L_0^{-1} E_0)} \\ &= \frac{(E_0, E_0)}{(E_0, K \omega_0^{-2} E_0)} = \underline{\omega}^2. \end{aligned}$$

<sup>5</sup> D. A. Taggart and F. W. Schott, "Ferrite-Filled Cavity Resonator," Applied Science Research 25, November 1971, page 38.

<sup>6</sup> See the penultimate sentence of the first paragraph of the introduction.

$$\begin{aligned}\text{Also } \omega_0^2 &= \frac{(E_0, L_0 E_0)}{(E_0, E_0)} > \frac{(E_0, L E_0)}{(E_0, E_0)} = \frac{(E_0, \frac{1}{K} L_0 E_0)}{(E_0, E_0)} \\ &= \frac{(E_0, \frac{1}{K} \omega_0^2 E)}{(E_0, E_0)} = \bar{\omega}^2.\end{aligned}$$

Finally, we prove  $\underline{\omega} \leq \bar{\omega}$  using the Schwartz inequality:

$$(E_0, E_0)^2 = \left( \sqrt{K} E_0, \frac{1}{\sqrt{K}} E_0 \right)^2 \leq \| \sqrt{K} E_0 \|^2 \cdot \| \frac{1}{\sqrt{K}} E_0 \|^2$$

$$\text{so that } \frac{(E_0, E_0)}{(\sqrt{K} E_0, \sqrt{K} E_0)} \leq \frac{(\frac{1}{\sqrt{K}} E_0, \frac{1}{\sqrt{K}} E_0)}{(E_0, E_0)}.$$

Theorem 2. If  $\omega$  is the lowest resonant frequency,  $\omega_1 \leq \omega \leq \bar{\omega}$ .

Proof. Because  $\omega^2$  is minimal for  $L$  and  $E_0$  belongs to  $D$ , we have

$$(E_0, (L - \omega^2) E_0) \geq 0 \quad \text{which implies } \omega^2 \leq \frac{(E_0, L E_0)}{(E_0, E_0)} = \bar{\omega}^2.$$

Similarly, we have  $\omega_1^2$  maximal for  $L_1^{-1}$  and

$$(E_1, (L_1^{-1} - \omega_1^2) E_1) \leq 0$$

$$\text{so } \omega_1^2 \leq \frac{(E_1, E_1)}{(E_1, L^{-1} E_1)} = \underline{\omega}^2.$$

Remark. The result  $\omega \leq \bar{\omega}$  is often called Rayleigh's principle.

Figures G1 through G5 show how  $\underline{x}/x_0$  and  $\bar{x}/x_0$  vary as a spherical cavity is filled with liquid nitrogen in the presence of a steady gravitational field.<sup>7</sup> Preliminary data indicate that the true value lies between these two estimates for the fundamental mode.

The Rayleigh-Ritz method is a refinement of the foregoing which also produces an upper bound for  $\omega$ . We will consider here only the case of the fundamental mode. The force of Theorem 2 is that

$$\omega^2 \leq \frac{(F, LF)}{(F, F)}$$

when  $F$  is any continuous vector field on the cavity which is normal to the walls and twice differentiable in the interior. Theorem 2 states this for the case in which  $F$  is the electric field of the fundamental mode for an empty cavity. We now consider the case in which  $F$  is a finite sum

$$F = \sum_{n=1}^N C_n F_n$$

Where  $F_n$  is the electric field of the  $n$ th mode in an empty cavity,  $\|F_n\| = 1$ ,  $C_n$  is a complex number to be determined later, and  $N$  is a positive integer. The preceding inequality then leads to

$$\omega^2 \leq \frac{\sum_{n=1}^N \sum_{n=1}^N C_n \bar{C}_n \omega_n^2 (F_n, \frac{1}{K} F_n)}{\sum_{n=1}^N |C_n|^2}$$

where  $\omega_n$  is the angular resonant frequency of the  $n$ th mode in the empty cavity. Then

<sup>7</sup> These results have been calculated numerically by techniques described in Appendix H.

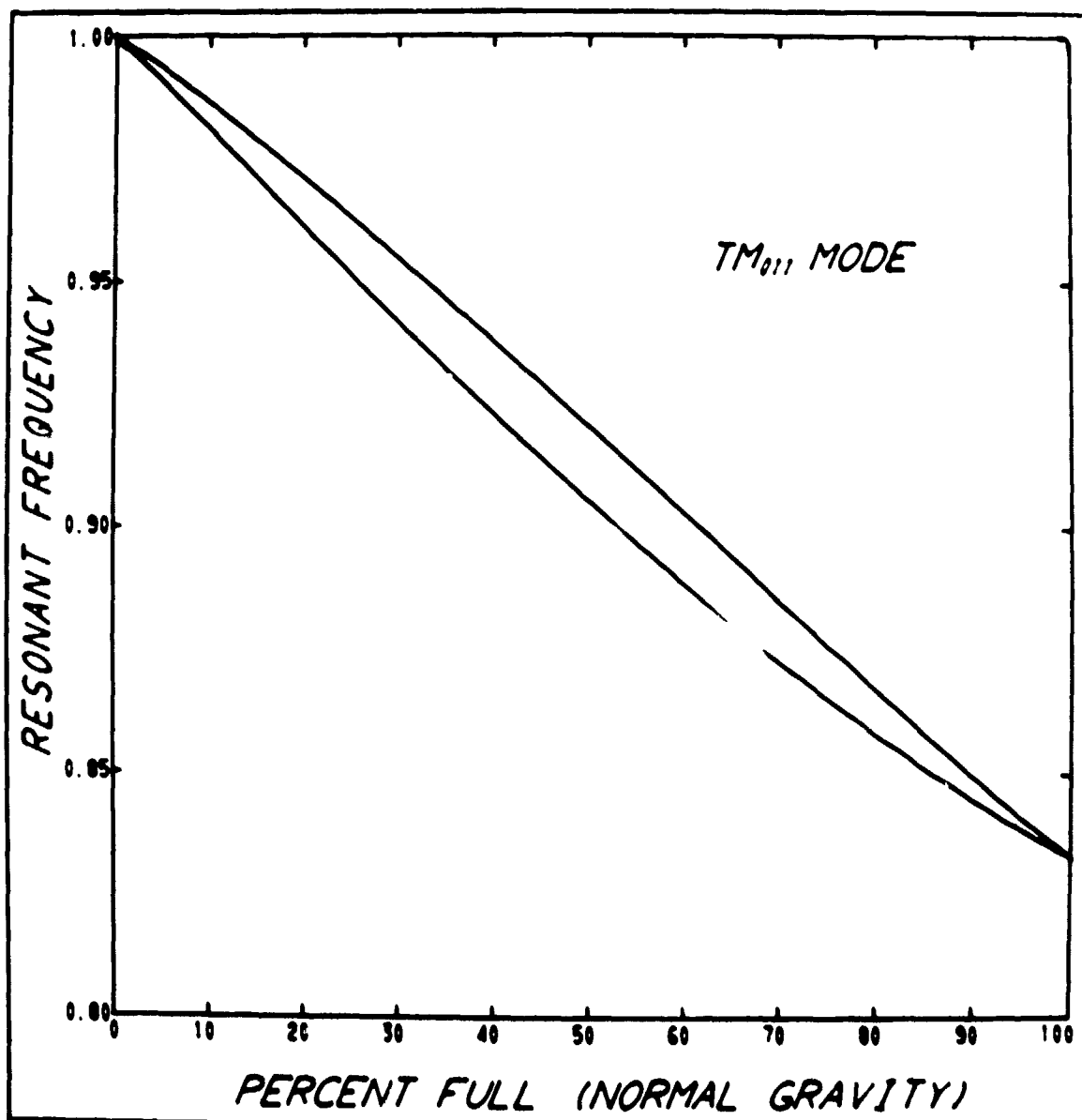


Figure G1

Upper and lower Rayleigh approximations to the normalized frequency  $\omega/\omega_0$  as a function of fill fraction, for the  $TM_{011}$  mode using liquid nitrogen.

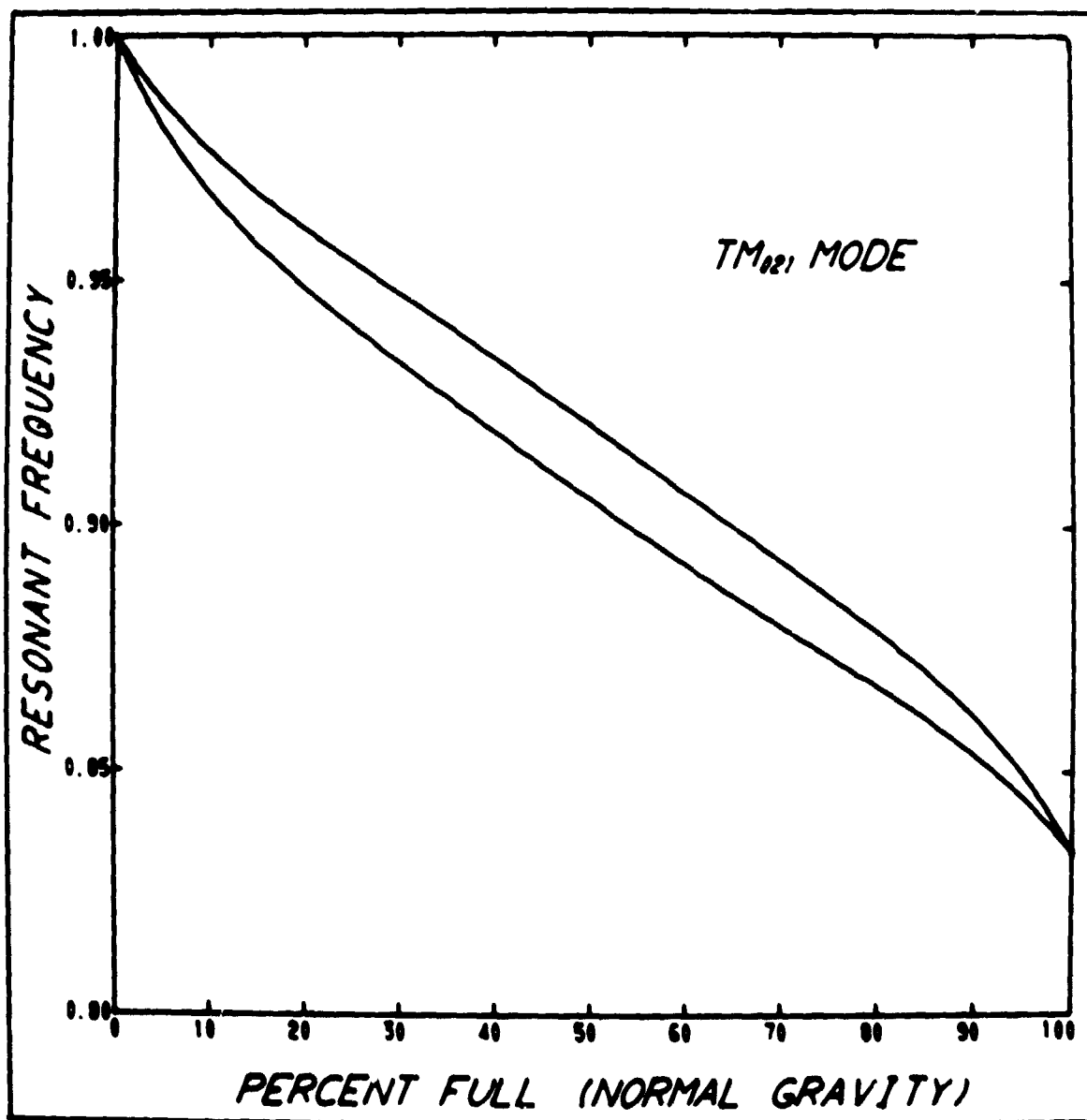


Figure G2

Upper and lower Rayleigh approximations to the normalized frequency  $\omega/\omega_0$  as a function of fill fraction, for the  $TM_{021}$  mode using liquid nitrogen.



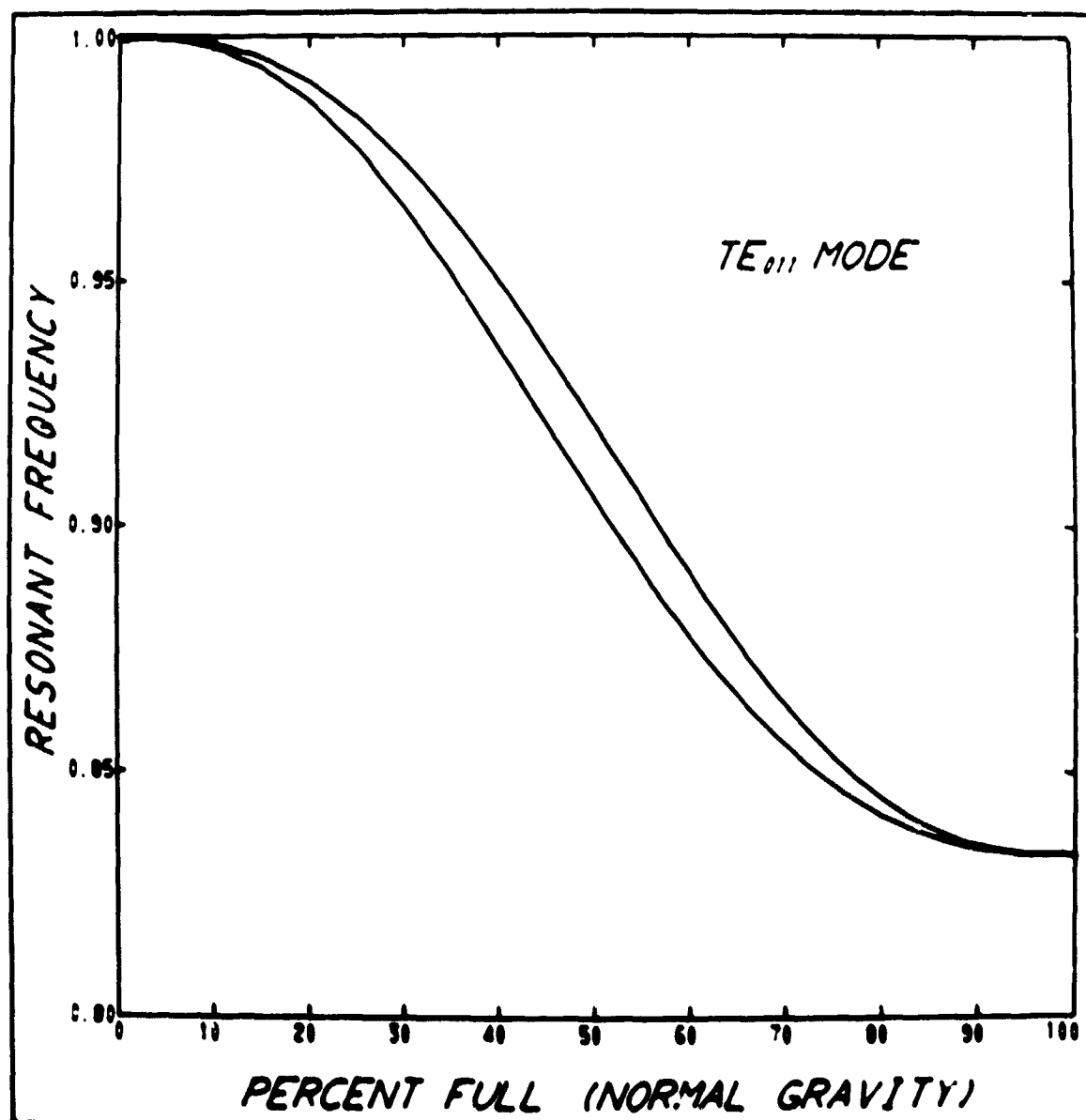


Figure G3

Upper and lower Rayleigh approximations to the normalized frequency  $\omega/\omega_0$  as a function of fill fraction, for the  $TE_{011}$  mode using liquid nitrogen.

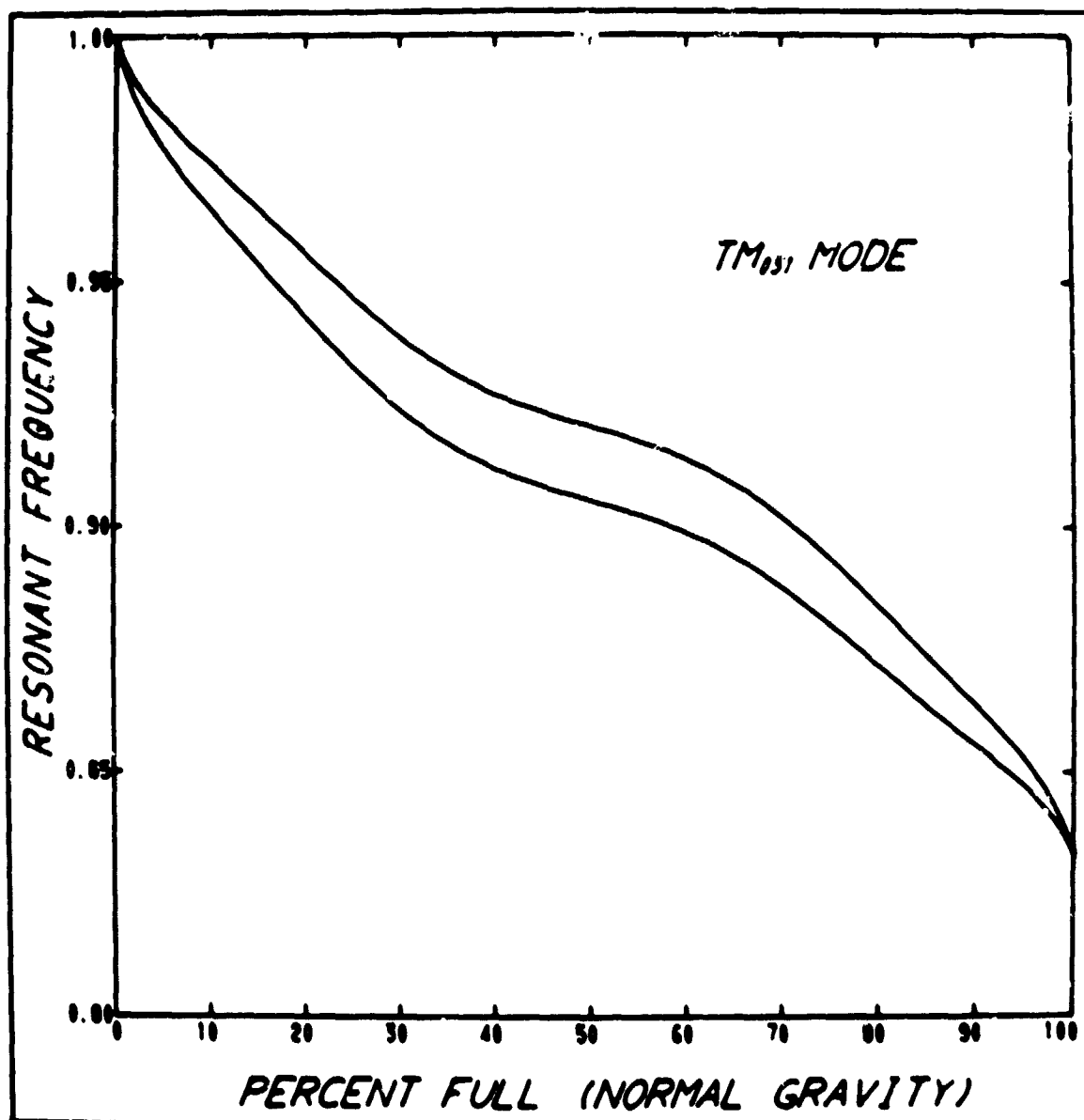


Figure G4

Upper and lower Rayleigh approximations to the normalized frequency  $\omega/\omega_0$  as a function of fill fraction  $n$ , for the  $TM_{031}$  mode using liquid nitrogen.

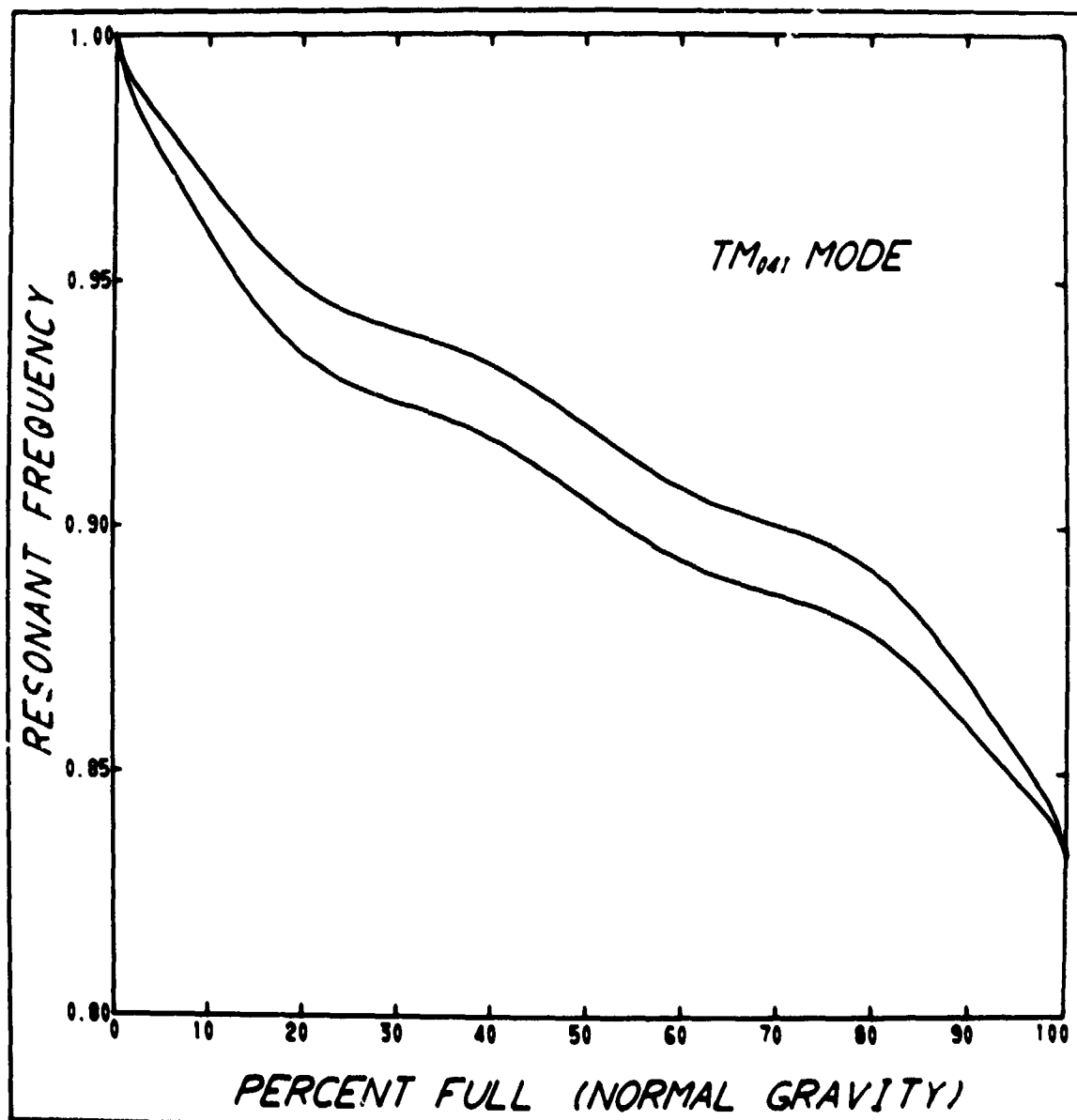


Figure G5

Upper and lower Rayleigh approximations to the normalized frequency  $\omega/\omega_0$  as a function of fill fraction, for the  $TM_{041}$  mode using liquid nitrogen.

$$\omega^2 \leq \min \sum_{n=1}^N \sum_{n=1}^N C_n \bar{C}_n \omega_n^2 (F_n, \frac{1}{K} F_n)$$

where the minimum is taken over all choices of  $C_1, C_2, \dots, C_N$  for which

$$\sum_{n=1}^N |C_n|^2 = 1$$

This is easily solved by Lagrange's method of multipliers, because the coefficients  $(F_n, \frac{1}{K} F_n)$  can be computed from results already obtained for the lower modes in the empty cavity. We propose not only to estimate  $\omega$  this way, but also to investigate the way this estimate depends on  $K$ .

#### The Moment Method

The moment method was pioneered by Temple<sup>8</sup>, elucidated by Kato<sup>9</sup> and generalized to higher order by Stackgold<sup>10</sup>. It derives its name from the set of numbers

$$m_n = \frac{(L^n E_0, E_0)}{(E_0, E_0)}, \quad n=1, 2, 3, \dots$$

called moments of the operator  $L$  with respect to the vector function  $E_0$ . Stackgold<sup>11</sup> showed that

<sup>8</sup> G. Temple, "The Theory of Rayleigh's Principle as Applied to Continuous Systems," Proceedings of the Royal Society of London, vol 119 (1928), p. 276-293.

<sup>9</sup> Tosio Kato, "On the Upper and Lower Bounds of Eigenvalues," Journal of the Physical Society of Japan, vol 4 (1949), pp. 334-339.

<sup>10</sup> Ivar Stackgold, The Use of Moments in Estimating the Spectrum, Department of the Navy, O. N. R. London Branch, August 1968, pp. 1-19, Accession Number A0839570.

<sup>11</sup> Lynn Taylor Wells, Extensions, Generalization, and Clarifications of Lower Bound Methods Applied to Eigenvalue Problems of Continuous Elastic Systems, Dissertation for the PhD at Ohio State Univ. (1970), pp. 66ff.

$$m_n - \frac{m_{2n} - m_n^2}{b^n - m_n} \leq \omega^{2n} < m_n + \frac{m_{2n} - m_n^2}{m_n - a^n}$$

where  $\omega$  is the angular resonant frequency of some empty cavity mode whose electric field is  $E_0$ . (The numbers  $a$ ,  $b$  will be defined shortly.) For our purposes it will be convenient to normalize the moments as follows:

$$M_n = \frac{m_n}{\omega_0^{2n}} = \frac{(L^n E_0, E_0)}{\omega_0^{2n} (E_0, E_0)}$$

so that the inequality becomes

$$M_n - \frac{M_{2n} - M_n^2}{\left(\frac{b}{\omega_0^2}\right)^n - M_n} \leq \left(\frac{\omega}{\omega_0}\right)^{2n} \leq M_n + \frac{M_{2n} - M_n^2}{M_n - \left(\frac{a}{\omega_0^2}\right)^n}$$

The real numbers  $a$ ,  $b$  are chosen as far apart as possible consistent with the condition that  $\omega^2$  be the only spectral point of  $L$  which lies strictly between  $a$ ,  $b$ . Then for the fundamental mode the best choice of  $a$  is  $-\infty$  and the best choice of  $b$  is the angular resonant frequency of the first harmonic if it were known. This leads to the simplification

$$M_n - \frac{M_{2n} - M_n^2}{(\gamma_n^2 - 1)M_n} \leq \left(\frac{\omega}{\omega_0}\right)^{2n} \leq M_n$$

for the fundamental mode, where

$$\gamma_n^2 = \left(\frac{b}{\omega_0^2}\right)^n \frac{1}{M_n}$$

Thus for the simplest estimate ( $n=1$ ) the upper bound is the upper Rayleigh estimate

$$M_1 = \frac{(\frac{1}{K} L_0 E_0, E_0)}{\omega_0^2(E_0, E_0)} = \frac{\int \frac{1}{K} E_0^2}{\int E_0^2} = \left(\frac{\bar{w}}{\omega_0}\right)^2,$$

and

$$\gamma_1^2 = \frac{b}{\bar{w}^2}.$$

For this simplest estimate we also need a value for

$$M_2 = \frac{(LE_0, LE_0)}{\omega_0^4(E_0, E_0)} = \frac{(\frac{1}{K} L_0 E_0, \frac{1}{K} L_0 E_0)}{\omega_0^4(E_0, E_0)} = \frac{\int \frac{1}{K^2} E_0^2}{\int E_0^2}.$$

Consider the special case of a spherical cavity half-full of liquid dielectric with the fundamental mode symmetrically oriented across the interface. Then

$$\int_{\text{LIQUID}} E_0^2 = \int_{\text{GAS}} E_0^2$$

so that for  $n = 1, 2$

$$M_n = \frac{\frac{1}{K_1^n} \int_{\text{LIQUID}} E_0^2 + \int_{\text{GAS}} E_0^2}{\int E_0^2} = \frac{1}{2} \left( \frac{1}{K_1^n} + 1 \right).$$

Then if the liquid is nitrogen ( $K_1 = 1.44$ ),

$$M_1 = 0.8472 \quad \text{and} \quad M_2 = 0.7411.$$

The data indicate the resonant frequencies of the lowest two modes have the ratio

$$\frac{700 \text{ MHz}}{510 \text{ MHz}} = 1.37$$

so

$$\gamma_1 = \frac{\sqrt{b}}{\omega_0} \frac{1}{\sqrt{M_1}} \leq \frac{1.37}{\sqrt{0.8472}} = 1.488 .$$

Hence our best possible estimate of this type is

$$\begin{aligned} \left( \frac{\bar{\omega}}{\omega_0} \right)^2 &= 0.847 \geq \left( \frac{\omega}{\omega_0} \right)^2 \geq 0.847 - \frac{0.7411 - 0.7178}{(1.488^2 - 1)0.8472} \\ &= 0.847 \frac{.0233}{1.028} \\ &= 0.824 \end{aligned}$$

$$\bar{\omega}/\omega_0 = 0.920 \geq \omega/\omega_0 \geq 0.908 \geq 0.906 = \underline{\omega}/\omega_0 .$$

If  $\gamma_1 = 1.3$  we obtain  $\omega/\omega_0 \geq 0.902$  and if  $\gamma_1 = 1.2$ ,  $\omega/\omega_0 \geq 0.886$  (see Fig. G6).

Oddly enough, the upper bound does not improve as  $n$  increases from 1 to 2:

$$\sqrt{M_1} = .920 \quad \text{and} \quad \sqrt{M_2} = .928 .$$

Calculating the lower bound corresponding to  $n = 2$  is much more difficult because

$$M_4 = \frac{(L^4 E_0, E_0)}{\omega_0^8 (E_0, E_0)} = \frac{(L^3 E_0, L E_0)}{\omega_0^8 (E_0, E_0)}$$

involves calculating higher powers of  $L$  operating on  $E_0$ .

The principle disadvantage of this class of methods is that it requires approximate knowledge of the next eigenvalue higher than the one being estimated. That is, we must estimate  $b$  or  $\gamma_n$ . But the above results for a spherical cavity suggest that the lower Rayleigh estimate is indeed a lower bound for the fundamental mode. This hypothesis should be investigated for a spherically symmetric distribution of dielectric material.

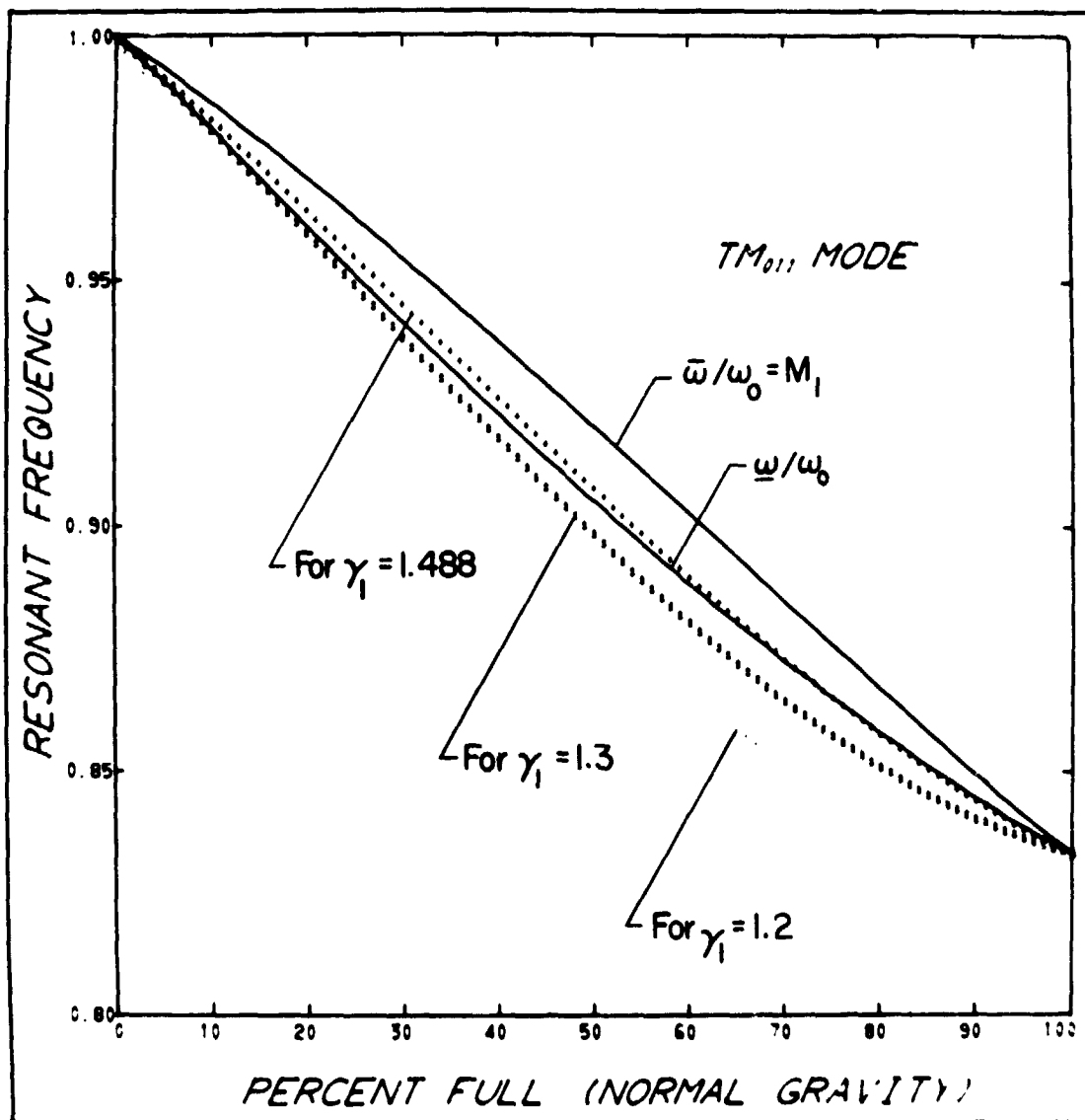


Figure G6 (Compare Figure G1)

Lower bounds for  $\omega/\omega_0$  from the first order method of moments, using three different values of  $\gamma_1$ . The solid lines are the Rayleigh upper and lower approximations.



### Green's Function Methods

Lemma. For any cavity  $V$  of diameter  $D$

$$\int_V \frac{d^3 R'}{(4\pi)^2 |R - R'|^2} \leq \frac{D}{4\pi} \quad (R \in V),$$

and for a spherical cavity  $V$  of diameter  $D$ ,

$$\int_V \frac{d^3 R'}{(4\pi)^2 |R - R'|^2} \leq \frac{D}{4\pi} \left( \frac{1 + \sqrt{2}}{4} \right) \quad (R \in V).$$

Proof. Make a change of variable

$$S = R' - R.$$

Since  $R \in V$ , the point  $S = 0$  always belongs to the cavity.

Then (for any shape of cavity) as  $R'$  varies over the cavity,  $S$  varies over a region which is the cavity translated by the vector  $-R$ . So regardless of  $R$  the region of integration is contained in a sphere of radius  $D$  about  $-R$ . Hence

$$\int_{V-R} \frac{d^3 S}{(4\pi)^2 |S|^2} \leq \frac{1}{4\pi} \int_0^D \frac{s^2 ds}{s^2} \int_0^{4\pi} \frac{d\omega}{4\pi} = \frac{D}{4\pi}$$

where  $\omega$  denotes solid angle.

In the spherical case, refer to Figure G7, which is a cross-sectional view of the region of integration. If the  $R'$ -origin (marked by the vector  $-R$ ) is eccentric from the  $S$ -origin by a distance  $aD$ , then the cavity will be wholly contained in a pair of hemispheres centered at the  $S$ -origin and having radii  $(\frac{1}{2} + a)D$  and  $bD$ , where  $a^2 + b^2 = (1/2)^2$ . Then we have

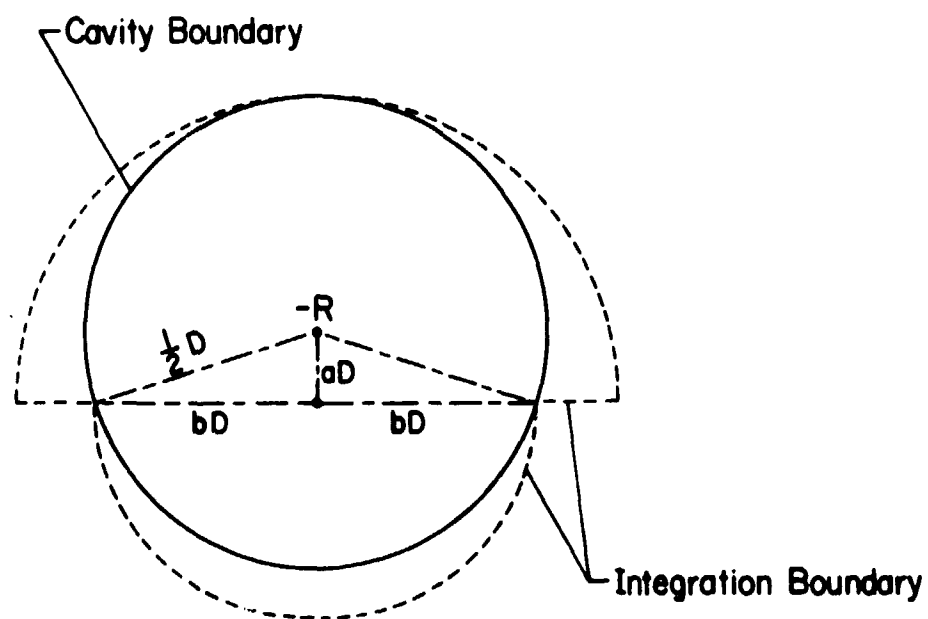


Figure G7

A cross-sectional view of the region of S - integration for the Lemma. The  $R'$  - origin is marked by the vector  $-R$ , and is eccentric from the S - origin by a distance  $aD$ .

$$\begin{aligned}
\int_{V-R} \frac{d^3 S}{(4\pi)^2 |S|^2} &\leq \frac{1}{4\pi} \int_0^{(\frac{1}{2}+a)D} \frac{s^2 ds}{s^2} \int_0^{2\pi} \frac{d\phi}{4\pi} + \frac{1}{4\pi} \int_0^{bD} \frac{s^2 ds}{s^2} \int_{2\pi}^{4\pi} \frac{d\phi}{4\pi} \\
&= \frac{1}{4\pi} \cdot \left(\frac{1}{2}+a\right) D \cdot \frac{1}{2} + \frac{1}{4\pi} \cdot bD \cdot \frac{1}{2} \\
&= \frac{D}{8\pi} \left(\frac{1}{2} + a + b\right) \\
&= \frac{D}{16\pi} (1 + 2a + 2b)
\end{aligned}$$

where  $(2a)^2 + (2b)^2 = 1$ . Let  $2a = \cos \theta$ ,  $2b = \sin \theta$  and maximize  $f(\theta) = 1 + \sin \theta + \cos \theta$ . Since  $f'(\theta) = \cos \theta - \sin \theta$  we have the maximum of  $f(\frac{\pi}{4}) = 1 + \sqrt{2}$  and the lemma is proved.

Theorem 3.  $L^{-1}$  is bounded on LD and is extended by the operator M, where

$$MF(R) = \int \frac{\mu \epsilon(R') F(R') d^3 R'}{4\pi |R-R'|}$$

Furthermore, M is bounded and

$$\|L^{-1}\| \leq \|M\| \leq \frac{1}{c^2} \left[ \int K^2(R') \left\| \frac{1}{4\pi |R-R'|} \right\|_R^2 d^3 R' \right]^{1/2} \leq \frac{\|K\|}{c^2} \sqrt{\frac{D}{4\pi}}$$

regardless of the distribution of  $\epsilon$  or the shape of the cavity. Here  $c$  is the speed of light and  $D$  is the greatest distance between any two points in the cavity,  $K = \epsilon/\epsilon_0$ , and  $\|K\| \left[ \int K^2 d^3 R' \right]^{1/2}$ .

Proof. First we compute LM using the identity

$\nabla \times \nabla \times = \nabla \nabla \cdot - \nabla^2$  and the fact that

$$\begin{aligned} -\nabla^2 M_1'(R) &= - \int \frac{\mu \epsilon(R') F(R')}{4\pi} \nabla^2 \left( \frac{1}{|R - R'|} \right) d^3 R' \\ &= - \int \frac{\mu \epsilon(R') F(R')}{4\pi} \left[ -4\pi \delta(R - R') \right] d^3 R' \\ &= \mu \epsilon(R) F(R), \end{aligned}$$

where  $\delta$  is the Dirac distribution.

Represent the  $m$ th Cartesian component of  $F$  by  $F_m$ , set  $g(R, R') = \frac{1}{4\pi |R - R'|}$ , and show differentiation by subscripts, referring to primed coordinates. Then we have the following identity in Cartesian tensors:

$$(\mu \epsilon F_m g, n)_{,n} = (\mu \epsilon F_m)_{,n} g, n + \mu \epsilon F_m g, n n.$$

Now integrate both sides over the cavity volume in primed coordinates and apply the divergence theorem to the left hand side to obtain

$$\int \mu \epsilon F_m g, n d^3 R' = \int (\mu \epsilon F_m)_{,n} g, n d^3 R' + \int \mu \epsilon F_m g, n n d^3 R'.$$

The left hand side vanishes if the surface of integration is described just outside the (lossless) cavity. The result expressed in vector notation is

$$0 = \int (\nabla' \cdot \mu \epsilon F) \nabla' g d^3 R' + \int \mu \epsilon F \cdot \nabla' \nabla' g d^3 R'$$

where  $\nabla'$  is the nabla operating on primed coordinates.

Using this last result it is easy to show that

$$\begin{aligned}\nabla \nabla \cdot MF(R) &= \int \mu \epsilon F \cdot \nabla \nabla g \\ &= \int \mu \epsilon F \cdot \nabla' \nabla' g \\ &= - \int (\nabla' \cdot \mu \epsilon F) \nabla' g ,\end{aligned}$$

and it follows that

$$\nabla \times \nabla \times MF(R) = \mu \epsilon F - \int (\nabla' \cdot \mu \epsilon F) \nabla' g$$

or 
$$LMF(R) = F - \frac{1}{\epsilon} \int (\nabla' \cdot \epsilon F) \nabla' g .$$

We see that for every  $G$  in  $D$ ,

$$\nabla' \cdot \epsilon LG(R') = \nabla' \cdot \frac{1}{\mu} \nabla' \times \nabla' \times G(R') = 0 ,$$

whence  $LMLG = LG$ . But the boundary value problem for  $L$  has a unique solution, so  $L$  has a trivial kernel and  $L^{-1}$  exists. Thus

$$MLG = G ,$$

where  $G$  is arbitrary in  $D$ . This shows that  $M$  extends  $L^{-1}$ .

Since  $M$  extends  $L^{-1}$ , it will have at least as large a norm, and it now suffices to compute the norm of  $M$ .

$$\begin{aligned}\|MF(R)\|_2 &\leq \int \mu \epsilon(R') |F(R')| \left\| \frac{1}{4\pi |R-R'|} \right\|_2 d^3 R' \\ &\leq \left[ \int \mu^2 \epsilon^2(R') \left\| \frac{1}{4\pi |R-R'|} \right\|_2^2 d^3 R' \right]^{1/2} \left[ \int |F(R')|^2 d^3 R' \right]^{1/2}\end{aligned}$$

implies the first inequality asserted for  $\|M\|$ . The weaker bound on  $\|M\|$  comes from an estimate of

$$\frac{1}{4\pi |R-R'|} = \frac{1}{(4\pi)^2} \int \frac{d^3 R}{(R-R')^2} = \frac{D}{4\pi}$$

which leads easily to the second bound on  $M_1$  stated in the theorem.

Corollary. Suppose  $\varepsilon = \varepsilon_1 \geq \varepsilon_j$  for a fraction  $\alpha$  of the volume of the cavity and  $\varepsilon = \varepsilon_j$  for the remainder. Then the crude estimate of the theorem takes the form

$$\|M_1\| \leq \frac{D^2}{c^2} \left[ \frac{1 + \alpha (K_1^2 - 1)}{24} \right]^{1/2}$$

Proof.

$$\|K_1\|^2 = \int_{\varepsilon=\varepsilon_1} K_1^2 + \int_{\varepsilon=\varepsilon_j} 1^2$$

$$= [\alpha K_1^2 + (1-\alpha)] \frac{\pi}{6} D^3,$$

$$\text{so } \frac{D \|K_1\|^2}{4\pi} = D^4 \left[ \frac{\alpha K_1^2 + (1-\alpha)}{24} \right],$$

and the conclusion follows.

Theorem 4. For the fundamental mode,

$$\alpha \geq \frac{1}{\|L^{-n}\|^{1/2n}} > \frac{1}{\|M^n\|^{1/2n}}$$

for every positive integer  $n$ .

Proof.

$$\alpha^{2n} = \frac{(E, E)}{(L^{-n} E, E)},$$

$$\text{but } (L^{-n} E, E) \leq \|L^{-n} E\| \cdot \|E\| \leq \|L^{-n}\| \cdot \|E\|^2 \leq \|M^n\| \|E\|^2$$

by the Schwartz inequality and Theorem 3 above.

Thus  $\omega^2 \geq \frac{1}{L_{\parallel}^2} \geq \frac{1}{M_{\parallel}^2}$  and the assertion follows.

Remark. Since  $\lim_{n \rightarrow \infty} L_{\parallel}^{-n} 1/n$  is the spectral radius  $\rho(L_{\parallel}^{-1})$ , the reader may think the case  $n=1$  sufficient. But the more general estimate is given in the hope that it may facilitate computation.

Corollary.

$$\omega \geq \frac{c}{D} \left[ \frac{24}{1 + \alpha(K_1^2 - 1)} \right]^{1/4}$$

Proof. Immediate from the corollary to Theorem 3.

Application. For an empty spherical cavity 48 cm in diameter (any shape)  $\alpha=0$  and  $D=0.48$  meter, so

$$\frac{\omega}{2\pi} \geq \frac{\sqrt[4]{24} \times 300 \times 10^8 \text{ meter/sec}}{2\pi \times 0.48 \text{ meter}} = 220 \text{ MHz.}$$

But theoretical calculations<sup>12</sup> give an exact theoretical value of 543 MHz. and experimental verification using liquid nitrogen at atmospheric pressure gives 546 MHz. So our crude bound is quite crude. But we hope for a better result when

$$\left\| \frac{1}{4\pi(R-R')} \right\|$$

is calculated accurately as a function of  $R'$ .

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<sup>12</sup> See Appendix F, Table I.

A somewhat different group of techniques is associated with higher powers of  $M$ .

$$M^n F(R) = \int \dots \int \frac{K(R_1) \dots K(R_n) F(R_n) d^3 R_1 \dots d^3 R_n}{4\pi |R-R_1| \dots 4\pi |R_1-R_2| \dots 4\pi |R_{n-1}-R_n|}$$

$$M^n F(R) \leq \int \dots \int \omega^n \frac{1}{4\pi |R-R_1|} \frac{K(R_1) \dots K(R_n) |F(R_n)| d^3 R_1 \dots d^3 R_n}{4\pi |R_1-R_2| \dots 4\pi |R_{n-1}-R_n|}$$

$$\leq \omega^n \omega^n I_n^{1/2}(K) \left[ \int \dots \int F^2(R_n) d^3 R_1 \dots d^3 R_n \right]^{1/2}$$

$$\text{where } I_n(K) = \int \dots \int \frac{1}{4\pi |R-R_1|} \frac{K^2(R_1) \dots K^2(R_n) d^3 R_1 \dots d^3 R_n}{(4\pi)^2 |R_1-R_2|^2 \dots (4\pi)^2 |R_{n-1}-R_n|^2}$$

Continuing, we have for a sphere of diameter  $D$

$$\|M^n F(R)\| \leq \frac{1}{\omega^{2n}} I_n^{1/2}(K) \left[ \int d^3 R_1 \right]^{\frac{n-1}{2}} \left[ \int F^2(R_n) d^3 R_n \right]^{1/2}$$

If we call the last integral  $\|F\|^2$ , then for a sphere of diameter  $D$

$$\|M^n\| \leq \frac{1}{\omega^{2n}} I_n^{1/2}(K) \left[ \frac{\pi D^3}{6} \right]^{\frac{n-1}{2}}$$

$$\|M^n\|^{-\frac{1}{2n}} \leq \frac{1}{\omega^{4n}} I_n^{1/2}(K) \left[ \frac{6}{\pi D^3} \right]^{\frac{n-1}{4n}}$$

$$\|M^n\|^{-\frac{1}{4}} \leq \left( \frac{6}{\pi D^3} \right)^{\frac{1}{4}} \lim_n I_n^{1/2}(K)$$

To estimate  $I_n$ , observe that (Lemma to Theorem 3) for a spherical cavity of diameter  $D$ ,



$$\int \frac{K^2(R_n) d^3 R_n}{(4\pi)^2 (R_{n-1} - R_n)^2} \leq \frac{K_1^2 D}{4\pi} \left( \frac{1+\sqrt{2}}{4} \right)$$

$$\int \frac{K^2(R_{n-1}) K^2(R_n) d^3 R_{n-1} d^3 R_n}{(4\pi)^2 (R_{n-1} - R_n)^2 (4\pi)^2 (R_{n-1} - R_n)^2}$$

$$\leq \frac{K_1^2 D}{4\pi} \left( \frac{1+\sqrt{2}}{4} \right) \int \frac{d^3 R_{n-1}}{(4\pi)^2 (R_{n-1} - R_n)^2} \leq \left[ \frac{K_1^2 D}{4\pi} \left( \frac{1+\sqrt{2}}{4} \right) \right]^2$$

$$I_n(K) \leq \left[ \frac{K_1^2 D}{4\pi} \left( \frac{1+\sqrt{2}}{4} \right) \right]^{n-1} \int \frac{1}{4\pi |R - R_1|} K(R_1) d^3 R_1$$

$$\lim_n I_n^{1/n}(K) \leq \frac{K_1^2 D}{4\pi} \left( \frac{1+\sqrt{2}}{4} \right)$$

$$\omega \geq c \left[ \frac{6}{\pi D^3} \cdot \frac{4\pi}{K_1^2 D} \left( \frac{4}{1+\sqrt{2}} \right) \right]^{1/4} = \frac{C}{(D/2)\sqrt{K_1}} \left( \frac{6}{1+\sqrt{2}} \right)^{1/4}$$

$$\frac{\omega}{2\pi} \geq \frac{c}{\pi D \sqrt{K_1}} \left( \frac{6}{1+\sqrt{2}} \right)^{1/4} = \frac{300 \times 10^6}{3.14(.48)1.2} \times 1.254 = 208 \text{ MHz.}$$

And replacing  $K_1$  by 1 as an approximation for  $K$  in the estimate of  $I$  increases the answer by only a factor of 1.2 to the value 250 MHz. This suggests the crudeness of the estimate is due primarily to the way  $I$  was evaluated. Alternatively, set

$$J_n(K) = \int \cdots \int \frac{d^3 R_1 \cdots d^3 R_n}{(4\pi)^2 (R_1 - R_2)^2 \cdots (4\pi)^2 (R_{n-1} - R_n)^2}$$

and observe that for a sphere of diameter  $D$  (see Lemma).

$$I_n(K) \leq \frac{D}{4\pi} \left( \frac{1+\sqrt{2}}{4} \right) K_1^{-1/2} J_n(K)$$

$$c \left[ \frac{D}{4\pi} \left( \frac{1+\sqrt{2}}{4} \right) \right]^{-1/4n} K_1^{-1/2} J_n^{-1/4n}(K) \left[ \frac{6}{-D^3} \right]^{-\frac{n-1}{4n}}$$

$$\frac{1}{2\pi} \geq \frac{c}{2\pi/K_1} \left( \frac{6}{-D^3} \right)^{1/4} \lim_n J_n^{-1/4n}(K).$$

One final approach in this vein is only heuristic at this point. The formula for  $M^n F(R)$  is an integral whose integrand is overwhelmingly important when the variables

$$R, R_1, R_2, \dots, R_n$$

have all nearly the same value. This suggests the approximation

$$M^n F(R) \approx \mu^n \epsilon_0^n K^n(R) F(R) W_n(R)$$

$$\text{where } W_n(R) = \int \dots \int \frac{d^3 R_1 \dots d^3 R_n}{4\pi |R-R_1| 4\pi |R_1-R_2| \dots 4\pi |R_{n-1}-R_n|}.$$

This strictly positive function can be estimated by the technique of the lemma:

$$\begin{aligned}
\int_V \frac{d^3 R^1}{4\pi |R - R^1|} &= \int_{R+V} \frac{d^3 S}{4\pi S} \\
&\leq \int_0^{(\frac{1}{2}+a)D} |S| d|S| \int_0^{2\pi} \frac{d\Omega}{4\pi} + \int_0^{bD} |S| d|S| \int_{2\pi}^{4\pi} \frac{d\Omega}{4\pi} \\
&= \frac{1}{2} \left( \frac{1}{2} + a \right)^2 D^2 \cdot \frac{1}{2} + \frac{1}{2} b^2 D^2 \cdot \frac{1}{2} \\
&= \frac{1}{4} D^2 \left[ \left( \frac{1}{2} + a \right)^2 + b^2 \right] \\
&= \frac{1}{4} D^2 \left[ \frac{1}{4} + a + a^2 + b^2 \right] \\
&= \frac{1}{4} D^2 \left[ \frac{1}{4} + a + \frac{1}{4} \right] \\
&\leq \left( \frac{D}{2} \right)^2
\end{aligned}$$

because  $a \leq \frac{1}{2}$ .

Therefore  $W_n(R) \leq \left( \frac{D}{2} \right)^{2n}$ .

And since

$$\begin{aligned}
\omega^{2n} &= \frac{(E, E)}{(M^n E, E)} \approx \frac{(E, E)}{(c^{-2n} K^n W_n E, E)} \\
\omega &\approx \frac{c(E, E)^{1/2n}}{(K^n W_n E, E)^{1/2n}}
\end{aligned}$$

But for a two phase system

$$\begin{aligned}
 (K^n W_n E, E)^{1/n} &= \left[ K_1^n \int_{\text{LIQUID}} W_n E^2 + \int_{\text{GAS}} W_n E^2 \right]^{1/n} \rightarrow K_1^{1/2} \\
 &\leq \left[ K_1^n \left( \frac{D}{2} \right)^{2n} \int_{\text{LIQUID}} E^2 + \left( \frac{D}{2} \right)^{2n} \int_{\text{GAS}} E^2 \right]^{1/2n}
 \end{aligned}$$

and this upper bound approaches  $K_1^{1/2} \left( \frac{D}{2} \right)$  as  $n$  increases without bound. Hence

$$\omega \geq \frac{c}{(D/2) \sqrt{K_1}}.$$

So for an empty cavity

$$\frac{\omega}{2\pi} = \frac{300 \times 10^8}{2\pi(0.24)1.2} = 166 \text{ MHz.}$$

Whether the validity of this approach can be established is unknown, and even if it can be, there seems less likelihood of getting a good estimate for  $W_n(R)$  than for  $J_n(K)$ .

#### Recommendations for Further Study

1. Compute the Rayleigh estimates for several of the lower modes in the case of a spherically symmetric distribution of liquid.
2. Compute the bounds

$$M_n - \frac{M_{2n} - M_n^2}{\left( \frac{b}{\omega_0^2} \right)^n - M_n} \leq \left( \frac{\omega}{\omega_0} \right)^2 \leq M_n + \frac{M_{2n} - M_n^2}{M_n - \left( \frac{a}{\omega_0^2} \right)^n}$$

for  $n = 1, 2$  in several of the lower modes for both normal gravity fill and spherically symmetric distributions of liquid.

3. Investigate the Green's function method by computing  $\|M\|$ ,  $J_n(K)$ ,  $W_n(K)$  accurately.

## APPENDIX H

### NUMERICAL ANALYSIS OF THE SPHERICAL CAVITY FOR THE LOWEST ORDER MODES

#### Electric Field Contours

Equation (F-22) was used to generate the graphs plotted by sub-routine TMTEPLOT. These are plots of the electric field contours,  $|E(r, \theta)| = |E|$ , for the  $TM_{011}$ ,  $TM_{021}$ ,  $TM_{031}$ ,  $TM_{041}$ , and  $TE_{011}$  modes. Choosing  $C_{n1}$  as imaginary ( $n = 1, 2, 3, 4$ ) in (F-22) the equations used in TMORTE for the TM modes became

$$g = z j_n(z) P_n(\cos \theta) \quad (H-1)$$

$$z = u_{n1} \frac{r}{b} \text{ for } n = 1, 2, 3, 4$$

and  $u_{n1}$  are the eigen values of the different modes. (The outside radius,  $b$ , is normalized to 1.0.  $P_n$  and  $j_n$  are the Legendre and spherical Bessel functions, respectively.) With

$$|E| \equiv \sqrt{E_r^2 + E_\theta^2 + E_\phi^2}$$

being the magnitude of the electric field, and using (H-1), the components become

$$\begin{aligned} E_r &= \frac{-n(n+1)}{r^2} \left[ z j_n(z) P_n(\cos \theta) \right] \\ &= \frac{-n(n+1)}{z} \left( u_{n1} \right)^2 j_n(z) P_n(\cos \theta) \end{aligned}$$

Also

$$E_{\theta} = \frac{1}{r} \frac{\partial^2}{\partial r \partial \theta} \left[ z j_n(z) P_n(\cos \theta) \right]$$

$$= \frac{1}{r} \frac{d}{d\theta} \left[ P_n(\cos \theta) \right] \left\{ z j_{n-1}(z) - n j_n(z) \right\} \frac{dz}{dr}$$

and

$$E_{\varphi} = 0.$$

The analogous equations for the TE modes are

$$f = u_{n1} j_n(z) P_n(\cos \theta)$$

The magnitude of the electric field becomes  $|E| \equiv \sqrt{E_{\varphi}^2}$ .

The components

$$E_r = E_{\theta} = 0$$

and

$$E_{\varphi} = u_{n1} j_n(z) \frac{d}{d\theta} \left[ P_n(\cos \theta) \right].$$

The plots of  $|E|$  for the first six modes are shown in figures 8 through 13 in the body of this report. Before plotting,  $|E|$  was scaled by its largest value and normalized to 10. Hence, the contours are numbered 1, 2, ..., 10.

#### Upper and Lower Raleigh Approximations for the Normal Fill Geometry.

The Resonant Frequency vs. Percent Full (Normal Gravity) graphs generated by subroutine LINPT were obtained numerically integrating  $|E|^2$  over different portions of the resonating spherical cavity for the various modes. The graphs display the upper and lower

Rayleigh approximations for the true solution. Working in spherical coordinates, volume is a function of  $(r, \theta, \varphi)$ ,

$$\text{ie, } V = f(r, \theta, \varphi).$$

However, assuming symmetry in the  $\varphi$  direction,  $V = 2\pi g(r, \theta)$ .

Hence, the integrals for the different portions of the resonating cavity become

$$\begin{aligned} G \left[ E(V_1) \right] &= 2\pi \int_0^{\theta_k} \int_0^R |E|^2 r^2 \sin \theta \, dr \, d\theta \\ &+ 2\pi \int_{\theta_k}^{\pi} \int_0^{-H/\cos \beta} |E|^2 r^2 \sin \beta \, dr \, d\beta \end{aligned}$$

where

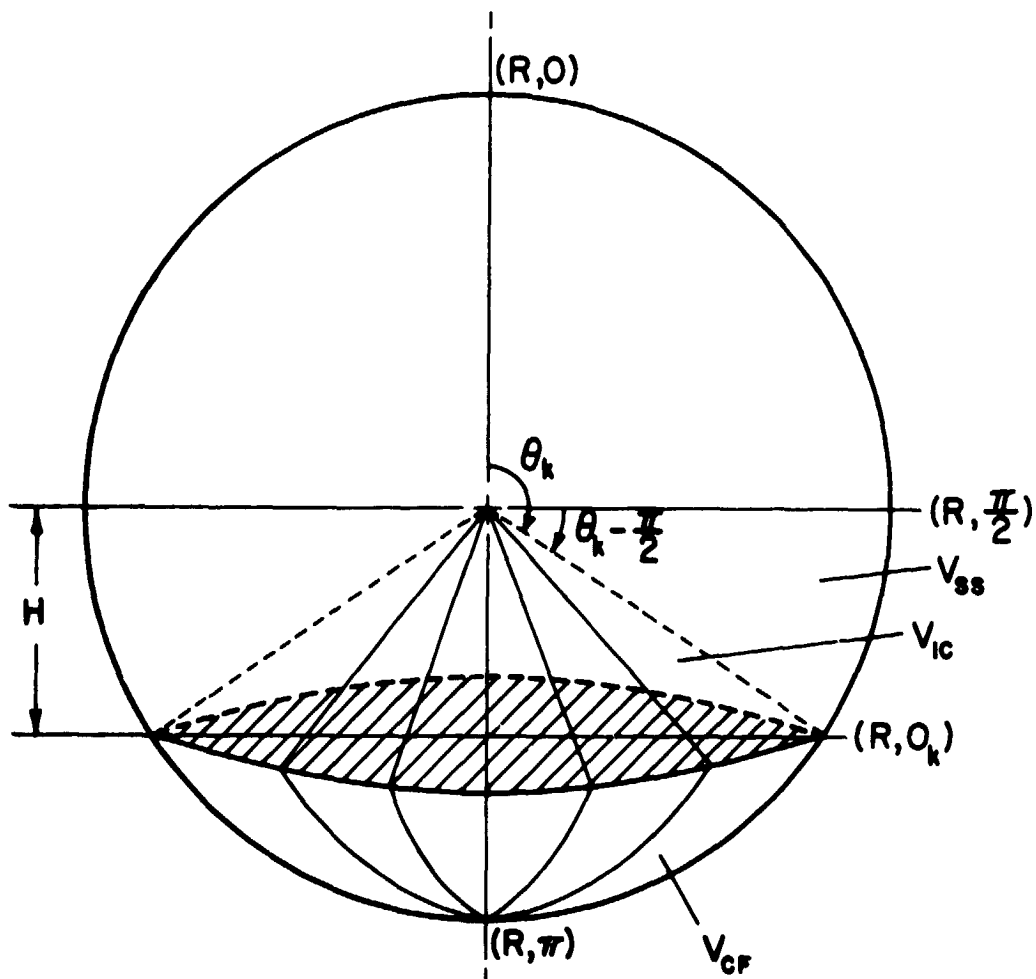
$$\frac{\pi}{2} \leq \beta \leq \theta_k \text{ (see figure H1 for } \theta_k \text{)}$$

and

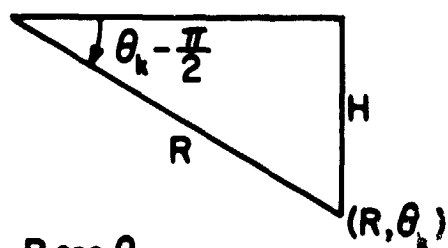
$$F \left[ E(V_{CF}) \right] = 2\pi \int_{\theta_k}^{\pi} \int_{-H/\cos \beta}^R |E|^2 r^2 \sin \beta \, dr \, d\beta.$$

G defines an integral of the field magnitude in the empty part ( $V_1$ ) of the cavity and F defines an integral of the field magnitude in the part of the cavity containing fluid ( $V_{CF}$ ). Figure H1 displays  $V_1$  and  $V_{CF}$ . The upper and lower approximations were then computed by evaluating.

$$\sqrt{\frac{\frac{1}{K} F + G}{C}} \quad \text{and} \quad \sqrt{\frac{C}{KF + G}}$$



$$V_1 = V_{ss} + V_{ic}$$



$$H = -R \cos \theta_k$$

Figure H1. Coordinate system for the normal fill geometry.



at steps of  $\Delta V_{CF} = 1\%$ . Here  $C = F + G$  and  $K$  is the dielectric constant of the fluid in the cavity.

Figures G1-G5 show the upper and lower Raleigh approximations for the first five spherical modes for the normal fill geometry.

### Computer Routines

Ten computer routines were developed and coded in Fortran to calculate the magnitude of the electric fields ( $|E(R, \theta)|$ ) and to integrate over the volume of the spherical cavity. These routines are as follows:

TMORTE, main program, calculates  $|E|$  for the transverse magnetic or transverse electric modes;

BJ, PN, and DPN are three function routines called by TMORTE to evaluate spherical Bessel functions of the 1st kind and Legendre polynomials and their derivatives, respectively;

TMTEPLOT, subroutine called by TMORTE, plots  $|E|$  for the transverse magnetic and transverse electric modes;

RCTOUR, subroutine called by TMTEPLOT, searches for specific contour values for TMTEPLOT to plot;

VOLUME, subroutine called by TMORTE, integrates  $|E|^2$  over the sphere to evaluate functions of the resonating frequency in the empty part and the full part of the spherical cavity;

RINTGL, subroutine called by VOLUME, performs that part of the integration along the radial lines,  $R$ , of the cavity;

LINPT, subroutine called by VOLUME, plots the upper and lower bounds of a Rayleigh approximation to the true solution for a Frequency vs. Percent Full graph;

INTRPL, subroutine called by VOLUME, interpolates between values of the percent full calculations to obtain required values.

A listing of the computer routines is given as follows:

C	PROGRAM TMORTE	TMT	1
		TMT	2
C	DIMENSION U(4), IFMT1(6), IFMT2(6), V(4)	TMT	3
		TMT	4
C	COMMON /DATA/ EMAGN(109,181),IRLAB(3),ICLAB(19),MODE,DELTA,DELTA,TMT	TMT	5
	1,NRINC,NTINC,NRIP1,NTIP1,PI,JJ(181),ESCALE	TMT	6
	COMMON /INDEX/ INDEX,EFSWT,GAMA1,GAMA2,GAMA3	TMT	7
C		TMT	8
	DATA (U=2.744,3.870,4.973,6.062), (B=1.), (PI=3.14159265), (IRLAB=18H	TMT	9
	1RADIAL DIRECTION 1), (ICLAB(1)=1H), (ICLAB(2)=1HT), (ICLAB(3)=1HH), (TMT	TMT	10
	2ICLAB(4)=1HE), (ICLAB(5)=1HT), (ICLAB(6)=1HA), (ICLAB(7)=1H), (ICLAB(TMT	TMT	11
	38)=1H), (ICLAB(9)=1HD), (ICLAB(10)=1HI), (ICLAB(11)=1HR), (ICLAB(12)=TMT	TMT	12
	41HE), (ICLAB(13)=1HC), (ICLAB(14)=1HT), (ICLAB(15)=1HI), (ICLAB(16)=1HTMT	TMT	13
	50), (ICLAB(17)=1HN), (ICLAB(18)=1H), (ICLAB(19)=1H), (INCR=9), (INCT=TMT	TMT	14
	610), (V=4.493,5.763,6.998,0.0), (IFMT1=47H(8H)TM MODE,2H (,12,1H)/5TMT	TMT	15
	77X,3A8/10X,13(13,6X))), (IFMT2=25H(1X,A1,13,1H0,2X,13E9.2/))	TMT	16
C		TMT	17
	INTEGER TMTE,EFSWT	TMT	18
	REAL KX	TMT	19
C		TMT	20
C	.....	TMT	21
C	ROUTINE TO CALCULATE COMPONENTS OF THE ELECTRIC FIELD OF THE	TMT	22
C	TRANSVERSE MAGNETIC MODE UNDER SPHERICAL SYMMETRY. EQUATION	TMT	23
C	(A-22) OF NBS REPORT 9793 BECOMES	TMT	24
C		TMT	25
C	G = P [COS(THETA)] * Z * J (Z)	TMT	26
C	N N	TMT	27
C		TMT	28
C	WHERE Z = U R/B.	TMT	29
C	N1	TMT	30
C		TMT	31
C		TMT	32
C	ALSO CALCULATES COMPONENTS OF THE ELECTRIC FIELD FOR THE	TMT	33
C	TRANSVERSE ELECTRIC MODE UNDER SPHERICAL SYMMETRY. THE	TMT	34
C	ANALOGOUS EQUATION OF (A-22) THEN IS	TMT	35
C		TMT	36
C	F = P [COS(THETA)] * K * J (Z)	TMT	37
C	N N N	TMT	38
C		TMT	39
C	WHERE Z = K R = V R/B AND B IS THE RADIUS OF THE SPHERE.	TMT	40
C	N1 N1	TMT	41
C	.....	TMT	42
C		TMT	43
C	R=1.0	TMT	44
C	ASSIGN THETA THE VALUE 180 DEGREES (IN RADIAN)	TMT	45
	THETA=PI	TMT	46
C		TMT	47
1	READ (60,19) TMTE,MODE,NRINC,NTINC,DFLDR,EFSWT,GAMA1,GAMA2,GAMA3	TMT	48
C		TMT	49
C	EFSWT = 1, CALCULATE E (AND NORMALIZE) EVERY TIME	TMT	50
C	2, CALCULATE E (AND NORMALIZE) AND WRITE E ON MAG TAPE	TMT	51
C	(LOG. UNIT NO. 1)	TMT	52
C	3, READ (NORMALIZED) E FROM MAG TAPE (LOG. UNIT NO. 1)	TMT	53
C		TMT	54

	IF (EOF.EQ.60) 18.2	TMT 55
C		TMT 56
2	N=MODE/10	TMT 57
	ROTATE=DFLDR*PI/180.0	TMT 58
	IF (TMTE.EQ.2RTE) GO TO 3	TMT 59
C	TM MODE PARAMETERS	TMT 60
	INDEX=1	TMT 61
	UX=U(N)	TMT 62
	IFMT1(1)=8H(8H1TM M	TMT 63
	GO TO 4	TMT 64
C	TE MODE PARAMETERS	TMT 65
3	INDEX=2	TMT 66
	UX=V(N)	TMT 67
	IFMT1(1)=8H(8H1TE M	TMT 68
4	KX=UX/B	TMT 69
	INCR=9	TMT 70
	INCT=10	TMT 71
	IF (NRINC.NE.12) GO TO 5	TMT 72
	INCR=1	TMT 73
	INCT=1	TMT 74
5	DELTAR=R/NRINC	TMT 75
	NRIP1=NRINC+1	TMT 76
	DELTAT=THETA/NTINC	TMT 77
	NTIP1=NTINC+1	TMT 78
	DO 6 I=1,NTIP1	TMT 79
	JJ(I)=I-1	TMT 80
6	CONTINUE	TMT 81
	IF (EFSWT.EQ.3) GO TO 15	TMT 82
C		TMT 83
C	---COMPONENTS OF THE ELECTRIC FIELD-----	TMT 84
	EMAX=0.0	TMT 85
	DO 12 IR=1,NRIP1	TMT 86
	R=(IR-1)*DELTAR	TMT 87
	Z=KX*R	TMT 88
	DZDR=KX	TMT 89
	DO 12 IT=1,NTIP1	TMT 90
	THETA=(IT-1)*DELTAT+ROTATE	TMT 91
	GO TO (7,10), INDEX	TMT 92
C	RADIAL (R) DIRECTION	TMT 93
7	IF (R.NE.0.0) GO TO 9	TMT 94
	IF (N.EQ.1) GO TO 8	TMT 95
	EMAGN(IR,IT)=0.0	TMT 96
	GO TO 12	TMT 97
8	EMAGN(IR,IT)=(2.0*KX*UX/(3.0*B))	TMT 98
	GO TO 11	TMT 99
9	ER=-N*(N+1)*(KX**2)*PN(COS(THETA),N)*BJ(Z,N)/Z	TMT 100
C		TMT 101
C	ANGULAR (THETA) DIRECTION	TMT 102
	ETHETA=DPN(COS(THETA),-SIN(THETA),N)*(DZDR/R)*(Z*BJ(Z,N-1)-N*BJ(Z,N))	TMT 103
	IN)	TMT 104
C		TMT 105
	EMAGN(IR,IT)=SQRT(ER**2+ETHETA**2)	TMT 106
	GO TO 11	TMT 107
C	TE MODE	TMT 108

10	EPHI=DPN(COS(THETA),-SIN(THETA),N)*KX*BJ(2,N)	TMT 109
C	ER=ETHETA=0.0	TMT 110
	EMAGN(IR,IT)=ABS(EPHI)	TMT 111
11	IF (EMAGN(IR,IT).GT.EMAX) EMAX=EMAGN(IR,IT)	TMT 112
12	CONTINUE	TMT 113
C		TMT 114
	PRINT IFMT1, MODE,IRLAB,(JJ(J),J=1,NRIP1,INCR)	TMT 115
	DO 13 IT=1,NTIP1,INCT	TMT 116
	I=IT/10+1	TMT 117
	IF (INCT.EQ.1) I=IT	TMT 118
	PRINT IFMT2, ICLAB(I),JJ(I),(EMAGN(J,IT),J=1,NRIP1,INCR)	TMT 119
13	CONTINUE	TMT 120
C		TMT 121
C	SCALE E ACCORDING TO LARGEST VALUE AND NORMALIZE TO 10.0	TMT 122
	ESCALE=10.0/EMAX	TMT 123
	DO 14 IR=1,NRIP1	TMT 124
	DO 14 IT=1,NTIP1	TMT 125
	EMAGN(IR,IT)=EMAGN(IR,IT)*ESCALE	TMT 126
14	CONTINUE	TMT 127
	CALL TMTEPLOT	TMT 128
	IF (EFSWT.EQ.1) GO TO 16	TMT 129
	WRITE (1) ((EMAGN(IR,IT),IR=1,NRIP1),IT=1,NTIP1),ESCALE	TMT 130
	GO TO 16	TMT 131
C		TMT 132
15	READ (1) ((EMAGN(IR,IT),IR=1,NRIP1),IT=1,NTIP1),ESCALE	TMT 133
C		TMT 134
16	PRINT IFMT1, MODE,IRLAB,(JJ(J),J=1,NRIP1,INCR)	TMT 135
	DO 17 IT=1,NTIP1,INCT	TMT 136
	I=IT/10+1	TMT 137
	IF (INCT.EQ.1) I=IT	TMT 138
	PRINT IFMT2, ICLAB(I),JJ(I),(EMAGN(J,IT),J=1,NRIP1,INCR)	TMT 139
17	CONTINUE	TMT 140
	PRINT 20	TMT 141
C		TMT 142
	CALL VOLUME	TMT 143
	GO TO 1	TMT 144
C		TMT 145
18	PRINT 21	TMT 146
	CALL EXIT	TMT 147
C		TMT 148
19	FORMAT (R2.13.215,F5.0,15.3E10.3)	TMT 149
20	FORMAT (1H1)	TMT 150
21	FORMAT (///11H END OF JOB)	TMT 151
	END	TMT 152
	SUBROUTINE VOLUME	VOL 1
C		VOL 2
C	SPECIAL ROUTINE TO EVALUATE (FUNCTIONS OF) VOLUMES CUT FROM	VOL 3
C	A SPHERE BY A PLANE PERPENDICULAR TO THE VERTICAL AXIS OF	VOL 4
C	SYMMETRY. HERE	VOL 5
C		VOL 6
C	V = F(R,THETA,PHI) = 2*PI*G(R,THETA).	VOL 7
C		VOL 8
	DIMENSION SVSSOE(1), SVSSE(180), SVICOE(1), SVICE(180), SVCFO(1), VOL	9
	15VCF(180), XTRA(3), SVCFOE(1), SVCPE(180), XTRAE(3), XU(1), X(30),VOL	10

	2 Y0(1), Y(30), U(1-1), DEG(51), FVCF0(1), FVCF(30), GV(101), GV(3VOL	11
	30), DFVCF(51), DGV(51), Y1(101), Y2(101), Y1LB1(101), Y1LB2(101),VOL	12
	4 Y1LB3(101)	VOL 13
C	COMMON /DATA/ LMAGN(109,181),IRLAB(3),ICLAB(19),MODE,DELTAR,DELTATVOL	14
	1,NRINC,NTINC,NRIP1,NTIP1,PI,JJ(181),ESCALE	VOL 15
	COMMON /INCLX/ INDEX,EFSWT,GAMA1,GAMA2,GAMA3	VOL 16
C	EQUIVALENCE (NRINC,NIR), (NRIP1,NIRF1), (NTINC,NIT), (NTIP1,NITP1)VOL	17
	1, EQUIVALENCE (VOL,TVOLE)	VOL 18
C	DATA (F38=0.375)	VOL 19
C	D2R=PI/180.0	VOL 20
	TBX=PI	VOL 21
	RBX=1.0	VOL 22
	TP138=2.0*PI*F38	VOL 23
	IPSWT=1	VOL 24
	TVOL=4.0*PI/3.0	VOL 25
C	RE-SCALE TO ORIGINAL EMAGN AND SQUARE	VOL 26
	RSCALE=1.0/ESCALE	VOL 27
	DO 1 IR=1,NRIP1	VOL 28
	DO 1 IT=1,NTIP1	VOL 29
	EMAGN(IR,IT)=(EMAGN(IR,IT)*RSCALE)**2	VOL 30
1	CONTINUE	VOL 31
C	SET THE INITIAL AND TERMINAL VALUES FOR R AND THETA	VOL 32
	RA=0.0	VOL 33
	RB=1.0	VOL 34
	KTHETA=90	VOL 35
	TA=PI/2.0	VOL 36
	TB=PI	VOL 37
	XYZ=0.0	VOL 38
C	CALCULATE NECESSARY INDICES FOR A SIMPSONS 3/8THS INTEGRATION	VOL 39
	JB=(RA/RBX)*NIR+1.01	VOL 40
	JMAX=(RB/RBX)*NIR+0.01	VOL 41
	IB=(TA/TBX)*NIT+1.01	VOL 42
	IE=(TB/TBX)*NIT+0.01	VOL 43
	IEP1=IE+1	VOL 44
C	SOLVE FOR THE VOLUME OF THE SPHERICAL SEGMENT, U,LE,THETA,LE,PIVOT	VOL 45
	ISWT=1	VOL 46
	DO 2 I=1,91	VOL 47
	CALL RINTGL (ISWT,I,JB,JMAX,XYZ,VSSSE)	VOL 48
	ISWT=3	VOL 49
	IM1=I-1	VOL 50
	SVSSE(IM1)=F38*VSSE*SIN(IM1*DELTAT)*DELTAT	VOL 51
2	CONTINUE	VOL 52
	VALU=0.0	VOL 53
	DO 3 I=1,90,3	VOL 54
	VALU=VALU+SVSSE(I-1)+3.0*(SVSSE(I)+SVSSE(I+1) SVSSE(I+2)	VOL 55
3	CONTINUE	VOL 56
		VOL 57
		VOL 58
		VOL 59
		VOL 60
		VOL 61
		VOL 62
		VOL 63
		VOL 64

	VOLSSF=TP138*VALU	VOL	65
	TVOLE=VOL SSE+VOL SSE	VOL	66
	IK=91	VOL	67
C		VOL	68
C	SOLVE FOR THE REMAINING VOLUMES, PI/2*L*THETA*LE.PI	VOL	69
	N=-1	VOL	70
	DO 10 NTHETA=KTHETA,IE,3	VOL	71
	N=N+1	VOL	72
	Y(30-N)=NTHETA	VOL	73
	H=-RB*COS(NTHETA*D2R)	VOL	74
	KTHP1=NTHETA+1	VOL	75
	IK=KTHP1	VOL	76
C		VOL	77
C	SOLVE FOR THE VOLUMES, NTHETA*LE*THETA*LL.PI	VOL	78
	DO 7 I=KTHP1,IEP1	VOL	79
	IM=I-1	VOL	80
	TIM1=SINI(IM*DELTAT)*DELTAT	VOL	81
	IF (NTHETA-90) 5,4,5	VOL	82
4	R=0.0	VOL	83
	GO TO 6	VOL	84
5	BETA=IM*D2R	VOL	85
	R=-H/COS(BETA)	VOL	86
6	JE=(R/RBX)*NIR+0.01+0.6	VOL	87
C	VOLUME OF THE SPHERICAL SEGMENT CONTAINING FUEL	VOL	88
	CALL RINTGL (4,I,JE+1,JMAX,VCF,VCFE)	VOL	89
	T38=F38*TIM1	VOL	90
	SVCF(IM)=T38*VCF	VOL	91
	SVCFE(IM)=T38*VCFE	VOL	92
7	CONTINUE	VOL	93
C		VOL	94
	VALU2=0.0	VOL	95
	VALUZE=0.0	VOL	96
	DO 8 I=KTHP1,IE,3	VOL	97
	VALU2=VALU2+SVCF(I-1)+3.0*(SVCF(I)+SVCF(I+1))+SVCF(I+2)	VOL	98
	VALUZE=VALUZE+SVCFE(I-1)+3.0*(SVCFE(I)+SVCFE(I+1))+SVCFE(I+2)	VOL	99
8	CONTINUE	VOL	100
	VOLCF=TP138*VALU2	VOL	101
	VOLCFE=TP138*VALUZE	VOL	102
C		VOL	103
	PCTF=VOLCF/TVOL	VOL	104
	VOL1=TVOLE-VOLCFE	VOL	105
C		VOL	106
	X(30-N)=PCTF*100.0	VOL	107
	FVCF(N)=VOLCFE	VOL	108
	GV(N)=VOL1	VOL	109
	GO TO (9,10), IPSWT	VOL	110
9	WRITE (61,21) NTHETA,VOL SSE,VOL1,VOLCFE,PCTF,N	VOL	111
10	CONTINUE	VOL	112
C		VOL	113
	X(1)=X(1)+1.0E-8	VOL	114
	X(2)=X(2)+5.0E-8	VOL	115
	X(3)=X(3)+10.0E-8	VOL	116
	DO 11 I=1,101	VOL	117
	U(I)=I-1	VOL	118

11	CONTINUE	VOL 119
	CALL INTRPL (61,31,X0,Y0,51,U,DEG)	VOL 120
	WRITE (61,22) (U(I),DEG(I),I=1,51)	VOL 121
	WRITE (61,23) MODE	VOL 122
	GV1(29)=GV1(29)+1.0E-8	VOL 123
	GV1(30)=GV1(30)+5.0E-8	VOL 124
	DO 12 I=1,31	VOL 125
	Y(I-1)=270.0-Y(I-1)	VOL 126
12	CONTINUE	VOL 127
	RK=1.0/1.44	VOL 128
	RKSQ=RK**2	VOL 129
	N1=N2=N3=0	VOL 130
	DO 16 I=1,51	VOL 131
	CALL INTRPL (61,31,Y0,V10,1,DEG(I),DGV1(I))	VOL 132
	CALL INTRPL (61,31,Y0,FVCF0,1,DEG(I),DFVCF(I))	VOL 133
	FI=I-1	VOL 134
	FM1=(RK*DFVCF(I)+DGV1(I))/VOL	VOL 135
	Y1(I)=SKFPGDC=SQRT(FM1)	VOL 136
	Y2(I)=RKFPDGC=SQRT(VOL/(1.44*DFVCF(I)+DGV1(I)))	VOL 137
	FM1SQ=FM1**2	VOL 138
	FM2=(RKSQ*DFVCF(I)+DGV1(I))/VOL	VOL 139
	QLB1=FM1-(FM2-FM1SQ)/((GAMA1**2-1.0)*FM1)	VOL 140
	FLB1=1.0E+50	VOL 141
	IF (QLB1.LT.0.0) GO TO 13	VOL 142
	N1=N1+1	VOL 143
	Y1LB1(I)=FLB1=SQRT(QLB1)	VOL 144
13	QLB2=FM1-(FM2-FM1SQ)/((GAMA2**2-1.0)*FM1)	VOL 145
	FLB2=1.0E+50	VOL 146
	IF (QLB2.LT.0.0) GO TO 14	VOL 147
	N2=N2+1	VOL 148
	Y1LB2(I)=FLB2=SQRT(QLB2)	VOL 149
14	QLB3=FM1-(FM2-FM1SQ)/((GAMA3**2-1.0)*FM1)	VOL 150
	FLB3=1.0E+50	VOL 151
	IF (QLB3.LT.0.0) GO TO 15	VOL 152
	N3=N3+1	VOL 153
	Y1LB3(I)=FLB3=SQRT(QLB3)	VOL 154
15	WRITE (61,24) FI,DFVCF(I),DGV1(I),SKFPGDC,RKFPDGC,FM2,FLB1,FLB2,FLB3	VOL 155
	183	VOL 156
16	CONTINUE	VOL 157
	WRITE (61,25) VOL	VOL 158
	WRITE (61,23) MODE	VOL 159
	DO 20 I=52,101	VOL 160
	FI=I-1	VOL 161
	XDGV1=VOL-DGV1(102-I)	VOL 162
	XDFVCF=VOL-DFVCF(102-I)	VOL 163
	FM1=(RK*XDFVCF+XDGV1)/VOL	VOL 164
	Y1(I)=SKFPGDC=SQRT(FM1)	VOL 165
	Y2(I)=RKFPDGC=SQRT(VOL/(1.44*XDFVCF+XDGV1))	VOL 166
	FM1SQ=FM1**2	VOL 167
	FM2=(RKSQ*XDFVCF+XDGV1)/VOL	VOL 168
	QLB1=FM1-(FM2-FM1SQ)/((GAMA1**2-1.0)*FM1)	VOL 169
	FLB1=1.0E+50	VOL 170
	IF (QLB1.LT.0.0) GO TO 17	VOL 171
	N1=N1+1	VOL 172

17	Y1LB1(I)=FLB1=SQRT(QLB1)	VOL 173
	QLB2=FM1-(FM2-FM1SQ)/((GAMA2**2-1.0)*FM1)	VOL 174
	FLB2=1.0F+50	VOL 175
	IF (QLB2.LT.0.0) GO TO 18	VOL 176
	N2=N2+1	VOL 177
	Y1LB2(I)=FLB2=SQRT(QLB2)	VOL 178
18	QLB3=FM1-(FM2-FM1SQ)/((GAMA3**2-1.0)*FM1)	VOL 179
	FLB3=1.0E+50	VOL 180
	IF (QLB3.LT.0.0) GO TO 19	VOL 181
	N3=N3+1	VOL 182
	Y1LB3(I)=FLB3=SQRT(QLB3)	VOL 183
19	WRITE (61,24) FI,XDFVCF,XDGV1,SKFPGDC,RKFPGDC,FM2,FLB1,FLB2,FLB3	VOL 184
20	CONTINUE	VOL 185
	WRITE (61,25) VOL	VOL 186
C		VOL 187
	XX=0.0	VOL 188
	CALL LINPT (XX,XX,XX,10.0,0.0,100.0,4.0,0.8,1.0,1.0,XX,XX)	VOL 189
	CALL LINPT (U,Y1,101,XX,XX,XX,XX,XX,XX,2.1,XX)	VOL 190
	CALL LINPT (U,Y2,101,XX,XX,XX,XX,XX,XX,2.1,XX)	VOL 191
	N1X=101-N1+1	VOL 192
	CALL LINPT (U(N1X),Y1LB1(N1X),N1,XX,XX,XX,XX,XX,XX,2.2,1K.)	VOL 193
	N2X=101-N2+1	VOL 194
	CALL LINPT (U(N2X),Y1LB2(N2X),N2,XX,XX,XX,XX,XX,XX,2.2,1R*)	VOL 195
	N3X=101-N3+1	VOL 196
	CALL LINPT (U(N3X),Y1LB3(N3X),N3,XX,XX,XX,XX,XX,XX,2.2,1R*)	VOL 197
	CALL LINPT (XX,XX,XX,XX,XX,XX,XX,XX,XX,3.0,XX,XX)	VOL 198
C		VOL 199
	RETURN	VOL 200
C		VOL 201
21	FORMAT (/14,3F10.5,2PF8.2,15)	VOL 202
22	FORMAT (F8.1,F8.2)	VOL 203
23	FORMAT (10H1M MODE (,12,1H)/3X,3HPCT,4X,6HF(VCF),5X,5HG(V1),3X,17VOL	VOL 204
	1HSQRT((F/K + G)/C),3X,19HSQRT(C/(K*F + G)) *,12X,2HM2,9X,3HLB1,9X,VOL	VOL 205
	23HLB2,9X,3HLB3/)	VOL 206
24	FORMAT (F6.1,F10.3,F10.3,2F16.6,8X,F17.6,3(4X,F8.6))	VOL 207
25	FORMAT (/9H * K=1.44/5H C=,F9.5)	VOL 208
	END	VOL 209
	SUBROUTINE RINTGL (ISWT,I,JB,JE,SUM,SUME)	RIN 1
C		RIN 2
C	ROUTINE TO INTEGRATE ALONG THE RADIAL LINES, R, INVOLVED	RIN 3
C	IN A TWO DIMENSIONAL INTEGRAL TO SOLVE FOR (FUNCTIONS OF)	RIN 4
C	VOLUMES OF SPHERICAL SECTORS OR SEGMENTS	RIN 5
C		RIN 6
	DIMENSION RO(1), R(108), REO(1), RE(108)	RIN 7
C		RIN 8
	COMMON /DATA/ EMAGN(109,181),IRLAR(3),ICLAB(19),MODE,DELTAR,DELTATRIN	RIN 9
	1,NRINC,NTINC,NRIP1,NTIP1,PI,JJ(181),ESCALE	RIN 10
C		RIN 11
	GO TO (1,3,11,3), ISWT	RIN 12
C		RIN 13
1	DO 2 J=1,109	RIN 14
	JM1=J-1	RIN 15
	R(JM1)=((JM1*DELTAR)**2)*DELTAR	RIN 16
2	CONTINUE	RIN 17



	GO TO 11	RIN 18
C		RIN 19
3	SUM=0.0	RIN 20
	IF (JB-JE) 4,4,11	RIN 21
4	IJE=(JE/3)*3	RIN 22
	IJB=((JB+2)/3)*3-2	RIN 23
	IJBX=IJB	RIN 24
	IF ((JB-IJB)-1) 7,5,6	RIN 25
5	SUM=SUM+1.5*(R(IJB)+R(IJB+1))	RIN 26
6	SUM=SUM+1.5*R(IJB+1)+R(IJB+2)	RIN 27
	IJBX=IJB+3	RIN 28
	IF (IJE-IJBX) 11,11,7	RIN 29
7	DO 8 J=IJBX,IJE,3	RIN 30
	SUM=SUM+R(J-1)+3.0*(R(J)+R(J+1))+R(J+2)	RIN 31
8	CONTINUE	RIN 32
	IF ((JE-IJE)-1) 11,10,9	RIN 33
9	SUM=SUM+1.5*(R(IJE+2)+R(IJE+1))	RIN 34
10	SUM=SUM+1.5*R(IJE+1)+R(IJE)	RIN 35
C		RIN 36
C		RIN 37
11	SUME=0.0	RIN 38
	IF (JB-JE) 12,12,20	RIN 39
12	IJE=(JE/3)*3	RIN 40
	IJB=((JB+2)/3)*3-2	RIN 41
	IJBX=IJB	RIN 42
	DO 13 J=IJB,109	RIN 43
	JM1=J-1	RIN 44
	RE(JM1)=R(JM1)*EMAGN(J,1)	RIN 45
13	CONTINUE	RIN 46
	IF ((JB-IJB)-1) 16,14,15	RIN 47
14	SUME=SUME+1.5*(RE(IJB)+RE(IJB+1))	RIN 48
15	SUMF SUME+1.5*RE(IJB+1)+RE(IJB+2)	RIN 49
	IJBX=IJB+3	RIN 50
	IF (IJE-IJBX) 20,20,16	RIN 51
16	DO 17 J=IJBX,IJE,3	RIN 52
	SUME=SUME+RE(J-1)+3.0*(RE(J)+RE(J+1))+RE(J+2)	RIN 53
17	CONTINUE	RIN 54
	IF ((JE-IJE)-1) 20,18,19	RIN 55
18	SUME=SUME+1.5*(RE(IJE+2)+RE(IJE+1))	RIN 56
19	SUME=SUME+1.5*RE(IJE+1)+RE(IJE)	RIN 57
C		RIN 58
20	RETURN	RIN 59
	END	RIN 60
	SUBROUTINE LINPT(XI,YI,NPT,NXD,FMNX,FMXX,NYD,FMY,FMYX,N,LS,ISYCH)	LIN 1
C		LIN 2
C		LIN 3
C	LINEAR PLOTTING ROUTINE	LIN 4
C		LIN 5
	COMMON /DD/ IN,IOR,IT,IS,IC,ICC,IX,IY	LIN 6
	COMMON /DDC/ LU,LUC,IFL	LIN 7
	COMMON /LABEL/ LABELX(5),LABELY(5),LSWTX,LSWTY	LIN 8
	COMMON /DATA/ EMAGN(109,181),IRLAB(3),ICLAB(19),MODE,DELTAR,DELTAT	LIN 9
	1,NRINC,NTINC,NRIPI,NTIPI,PI,JJ(181),ESCALE	LIN 10
	COMMON /INDEX/ INDEX,FSWT,GAMA1,GAMA2,GAMA3	LIN 11

C	DIMENSION XI(NPT), YI(NPT), ID(4)	LIN 12
C		LIN 13
	DATA (ID=32H RG PETERSON X3184 )	LIN 14
	DATA (IXMIN=100), (IYMIN=100), (LSWTX=2), (LSWTY=2)	LIN 15
C	DATA (LABELX=5(8H )), (LABELY=5(8H ))	LIN 16
	DATA (LABELX=40H PERCENT FULL (NORMAL GRAVITY) )	LIN 17
	DATA (LABELY=40H RESONANT FREQUENCY )	LIN 18
C		LIN 19
	INTEGER EFSWT	LIN 20
C		LIN 21
	GO TO (1,10,21), N	LIN 22
C		LIN 23
1	IOR=0	LIN 24
	IN=0	LIN 25
	IC=0	LIN 26
	IF (EFSWT.EQ.3) CALL DDINIT (4,ID)	LIN 27
	IT=0	LIN 28
C	DRAW A BOX AROUND PLOTTING AREA	LIN 29
	CALL DDBOX (0,1023,0,1023)	LIN 30
C	SPECIAL LABEL PLOTTED	LIN 31
	IT=1	LIN 32
	IS=3	LIN 33
	IX=675	LIN 34
	IY=800	LIN 35
	LABEL=8HTM MODE	LIN 36
	IF (INDEX.EQ.2) LABEL=8HTE MODE	LIN 37
	CALL DDTAB	LIN 38
	CALL DDTABNAB (1,LABEL,1)	LIN 39
	MODEX=1000+MODE	LIN 40
	ENCODE (8,23,LBMODE) MODEX	LIN 41
	IS=1	LIN 42
	IX=715	LIN 43
	IY=790	LIN 44
	CALL DDTAB	LIN 45
	CALL DDTABNAB (1,LBMODE,1)	LIN 46
C		LIN 47
	NY=NYD+1	LIN 48
	LTYDIV=900/NYD	LIN 49
	IYMAX=IYMIN+LTYDIV*NYD	LIN 50
	NX=NXD+1	LIN 51
	LTXDIV=900/NXD	LIN 52
	IXMAX=IXMIN+LTXDIV*NXD	LIN 53
C		LIN 54
	DRAW DIVISIONS OR TICKMARKS FOR X-AXIS	LIN 55
C		LIN 56
	IY=IYMAX	LIN 57
	IX=MINX=IXMIN	LIN 58
	CALL DDBP	LIN 59
	IY=IYMIN	LIN 60
	CALL DDVC	LIN 61
	DO 4 I=2,NX	LIN 62
	IY=IYMAX	LIN 63
	IXS=IX=IXMIN+(I-1)*LTXDIV	LIN 64
	CALL DDBP	LIN 65

	IYS=IY=IYMAX-8	LIN 66
	CALL DDVC	LIN 67
	IX=IXS	LIN 68
	IY=IYS	LIN 69
	IF (I.EQ.NX) GO TO 3	LIN 70
	GO TO (3,2), LSWTX	LIN 71
2	IY=IYMIN+8	LIN 72
	CALL DDBP	LIN 73
3	IY=IYMIN	LIN 74
	CALL DDVC	LIN 75
4	CONTINUE	LIN 76
C		LIN 77
C	DRAW DIVISIONS OR TICKMARKS FOR Y-AXIS	LIN 78
	IX=IXMAX	LIN 79
	IY=MINY=IYMIN	LIN 80
	CALL DDBP	LIN 81
	IX=IXMIN	LIN 82
	CALL DDVC	LIN 83
	DO 7 I=2,NY	LIN 84
	IX=IXMAX	LIN 85
	IYS=IY=IYMIN+(I-1)*LTYDIV	LIN 86
	CALL DDRP	LIN 87
	IXS=IX=IXMAX-8	LIN 88
	CALL DDVC	LIN 89
	IX=IXS	LIN 90
	IY=IYS	LIN 91
	IF (I.EQ.NY) GO TO 6	LIN 92
	GO TO (6,5), LSWTY	LIN 93
5	IX=IXMIN+8	LIN 94
	CALL DDBP	LIN 95
6	IX=IXMIN	LIN 96
	CALL DDVC	LIN 97
7	CONTINUE	LIN 98
C		LIN 99
C	NUMBER THE X-AXIS	LIN 100
	IS=1	LIN 101
	IT=0	LIN 102
	IOR=0	LIN 103
	FINCX=(FMXX-FMNX)/NXD	LIN 104
	DO 8 I=1,NX	LIN 105
	IYS=IY=IYMIN-15	LIN 106
	IM1=I-1	LIN 107
	IXS=IX=IXMAX-IM1*LTxDIV-32	LIN 108
	FNUMB=FMXX-IM1*FINCX	LIN 109
	ENCODE (8,24,FNUMBX) FNUMB	LIN 110
	CALL DDTAB	LIN 111
	CALL DDTABNAB (1,FNUMBX,1)	LIN 112
A	CONTINUE	LIN 113
C	NUMBER THE Y AXIS	LIN 114
	IS=1	LIN 115
	IOR=0	LIN 116
	FINCY=(FMXY-FMNY)/NYD	LIN 117
	DO 9 I=1,NY	LIN 118
	IX=60	LIN 119

	IM1=I-1	LIN 120
	IY=IYMAX-IM1*LTIDIV	LIN 121
	FNUMB=FMXY-IM1*FENCY	LIN 122
	ENCODE (8.25,FNUMBY) FNUMB	LIN 123
	CALL DDTAB	LIN 124
	CALL DDTABNAB (1,FNUMBY,1)	LIN 125
9	CONTINUE	LIN 126
C		LIN 127
C	LABEL X AND Y AXES	LIN 128
	IT=1	LIN 129
	IS=3	LIN 130
	IX=40	LIN 131
	IY=30	LIN 132
	CALL DDTAB	LIN 133
	CALL DDTABNAB (5,LABELX(1),1)	LIN 134
	IX=30	LIN 135
	IY=40	LIN 136
	IOR=1	LIN 137
	CALL DDTAB	LIN 138
	CALL DDTABNAB (5,LABELY(1),1)	LIN 139
C		LIN 140
	XMIN=FMNX	LIN 141
	XMAX=FMXX	LIN 142
	YMIN=FMNY	LIN 143
	YMAX=FMXY	LIN 144
	FMINX=IXMIN	LIN 145
	FMAXX=IXMAX	LIN 146
	FMINY=IYMIN	LIN 147
	FMAXY=IYMAX	LIN 148
	IS=0	LIN 149
	IT=0	LIN 150
	IOR=0	LIN 151
	GO TO 22	LIN 152
C		LIN 153
10	IXL=IYL=0	LIN 154
	JSWT=1	LIN 155
	GO TO (11,12), LS	LIN 156
C	PLOT VECTORS	LIN 157
11	CALL DDCONVEC	LIN 158
	GO TO 13	LIN 159
C	PLOT SYMBOLS OR CHARACTERS	LIN 160
12	ICC=ISYCH	LIN 161
	CALL DDSYMBOL	LIN 162
13	DO 20 I=1,NPT	LIN 163
	X=XI(I)	LIN 164
	Y=YI(I)	LIN 165
	IX=(X-XMIN)/(XMAX-XMIN)*(FMAXX-FMINX)+FMINX	LIN 166
	IY=(Y-YMIN)/(YMAX-YMIN)*(FMAXY-FMINY)+FMINY	LIN 167
	SLOPE=(IY-IYL)/(IX-IXL)	LIN 168
	IF (IX.GE.IXMIN) GO TO 14	LIN 169
	IY=SLOPE*(IXMIN-IXL)+IYL	LIN 170
	IX=IXMIN	LIN 171
	JSWT=2	LIN 172
	GO TO 15	LIN 173

14	IF (IX.LE.IXMAX) GO TO 15	LIN 174
	IY=SLOPE*(IXMAX-IXL)+IYL	LIN 175
	IX=IXMAX	LIN 176
	JSWT=2	LIN 177
15	IF (IY.GE.IYMIN) GO TO 16	LIN 178
	IY=IYMIN	LIN 179
	GO TO 17	LIN 180
16	IF (IY.LE.IYMAX) GO TO 18	LIN 181
	IY=IYMAX	LIN 182
17	IX=(IY-IYL)/SLOPE+IXL	LIN 183
	JSWT=2	LIN 184
18	IF (I.EQ.1.AND.JSWT.EQ.2) GO TO 19	LIN 185
	IF (IX.EQ.IXL.AND.IY.EQ.IYL) GO TO 20	LIN 186
	CALL DDXY	LIN 187
19	IXL=IX	LIN 188
	IYL=IY	LIN 189
20	CONTINUE	LIN 190
C		LIN 191
	CALL DDTAB	LIN 192
	GO TO 22	LIN 193
C		LIN 194
C	FRAME ADVANCE	LIN 195
21	CALL DDFR	LIN 196
C		LIN 197
22	RETURN	LIN 198
C		LIN 199
23	FORMAT (I3)	LIN 200
24	FORMAT (F4.0)	LIN 201
25	FORMAT (F4.2)	LIN 202
	END	LIN 203-
	SUBROUTINE TMTEPLOT	TMP 1
C		TMP 2
C	SPECIAL PLOTTING ROUTINE TO PLOT MAGNITUDES OF ELECTRIC FIELDS	TMP 3
C		TMP 4
	COMMON /DD/ IN, IOR, IT, IS, IC, ICC, IX, IY	TMP 5
	COMMON /DDC/ LU, LUC, IFL	TMP 6
	COMMON /DATA/ EMAGN(109,181), IRLAB(3), ICLAB(19), MODE, DELTAR, DELTAT	TMP 7
	1, NRINC, NTINC, NRIP1, NTIP1, PI, JJ(181), ESCALE	TMP 8
	COMMON /INDEX/ INDEX, EFSWT, GAMA1, GAMA2, GAMA3	TMP 9
	COMMON /SUB/ IRR(181), IXS(181), IYS(181), PID2, PIDNTI, R, NP	TMP 10
C		TMP 11
	DIMENSION LABEL(5), IXN(10), RAD(36), ID(4)	TMP 12
C		TMP 13
	DATA (ID=32HRG PETERSON X3184), (LABEL=40H R F MASS	TMP 14
	1GUAGING-PLOT OF E, TM MODE), (RAD=20(475.0), 477.0, 478.0, 480.0, 4	TMP 15
	281.0, 483.0, 485.0, 488.0, 490.0, 488.0, 485.0, 483.0, 481.0, 480.0, 478.0, 4	TMP 16
	377.0, 475.0)	TMP 17
C		TMP 18
	IOR=0	TMP 19
	IC=0	TMP 20
	IT=0	TMP 21
	CALL DDINIT (4, ID)	TMP 22
C		TMP 23
C	DRAW A BOX AROUND PLOTTING AREA	TMP 24

	IN=1	TMP	25
	CALL DDBOX (0,1023,0,1023)	TMP	26
C		TMP	27
C	PLUT A CONTINUOUS CIRCLE CENTERED AT RASTER COOR. (512,562)	TMP	28
C	WITH RADIUS = 450 RASTER UNITS	TMP	29
	IS=0	TMP	30
	CALL DDCONVEC	TMP	31
	PIDNTI=PI/NTINC	TMP	32
	PID2=PI/2.0	TMP	33
	DO 1 I=1,NTIP1	TMP	34
	T=(1-I)*PIDNTI	TMP	35
	TT=PID2-T	TMP	36
	IX=COS(TT)*450.0+512.0	TMP	37
	IY=SIN(TT)*450.0+532.0	TMP	38
	IXS(NTIP1-I+1)=1024-IX	TMP	39
	IYS(NTIP1-I+1)=IY	TMP	40
	CALL DDXY	TMP	41
1	CONTINUE	TMP	42
	DO 2 I=2,NTIP1	TMP	43
	IX=IXS(I)	TMP	44
	IY=IYS(I)	TMP	45
	CALL DDXY	TMP	46
2	CONTINUE	TMP	47
C	NUMBER CIRCLE EVERY 10 DEGREES	TMP	48
	DO 3 I=1,36	TMP	49
	IDEG=(I-1)*10	TMP	50
	TT=PID2-IDEG*PIDNTI	TMP	51
	COSTT=COS(TT)	TMP	52
	SINTT=SIN(TT)	TMP	53
	CALL DDCONVEC	TMP	54
	IX=COSTT*450.0+512.0	TMP	55
	IY=SINTT*450.0+532.0	TMP	56
	CALL DDXY	TMP	57
	IX=COSTT*458.0+512.0	TMP	58
	IY=SINTT*458.0+532.0	TMP	59
	CALL DDXY	TMP	60
	IX=COSTT*RAD(I)+512.0	TMP	61
	IY=SINTT*RAD(I)+532.0	TMP	62
	IDEGX=1000+IDEG	TMP	63
	ENCODE (8,13,NV) IDEGX	TMP	64
	CALL DDTAB	TMP	65
	CALL DDTABNAB (1,NV,1)	TMP	66
3	CONTINUE	TMP	67
C		TMP	68
C	LABEL PLOTTED	TMP	69
	IT=1	TMP	70
	IS=3	TMP	71
	IX=10	TMP	72
	IY=25	TMP	73
	LABEL(5)=8HM MODE	TMP	74
	IF (INDEX.EQ.2) LABEL(5)=8HE MODE	TMP	75
	CALL DDTAB	TMP	76
	CALL DDTABNAB (5,LABEL(1),1)	TMP	77
	MODEX=1000+MODE	TMP	78

	ENCODE (8,13,LBMODE) MODEX	TMP	79
	IS=1	TMP	80
	IX=794	TMP	81
	IY=15	TMP	82
	CALL DDTAB	TMP	83
	CALL DDTABNAB (1,LBMODE,1)	TMP	84
C		TMP	85
	IS=0	TMP	86
	NTID2=NTINC/2+1	TMP	87
C		TMP	88
C	SEARCH FOR CONTOURS	TMP	89
	DO 12 NC=1,10	TMP	90
	R=NC	TMP	91
C	ASSIGN BEGINNING INDICES, 2 FOR TM OR TE 11,21,31, AND 20 FOR 41	TMP	92
	DO 5 ITH=1,NTIP1	TMP	93
	IF (MODE.EQ.41) GO TO 4	TMP	94
	IRR(ITH)=2	TMP	95
	GO TO 5	TMP	96
4	IRR(ITH)=20	TMP	97
5	CONTINUE	TMP	98
C		TMP	99
	ITH=1	TMP	100
6	CALL RCTOUR	TMP	101
	ITHB=ITH	TMP	102
C		TMP	103
C	PLOT THE CONTOURS	TMP	104
	IF (NP.EQ.0) GO TO 10	TMP	105
	CALL DDCONVEC	TMP	106
	DO 7 I=1,NP	TMP	107
	IX=IXS(I)	TMP	108
	IY=IYS(I)	TMP	109
	CALL DDXY	TMP	110
7	CONTINUE	TMP	111
	IXR=1024-IXS(NP)	TMP	112
	IF (IXS(1).EQ.IXR) GO TO 8	TMP	113
	CALL DDBP	TMP	114
	CALL DDCONVEC	TMP	115
8	DO 9 I=1,NP	TMP	116
	NPMIP1=NP-I+1	TMP	117
	IX=1024-IXS(NPMIP1)	TMP	118
	IY=IYS(NPMIP1)	TMP	119
	CALL DDXY	TMP	120
9	CONTINUE	TMP	121
	CALL DDBP	TMP	122
C		TMP	123
10	DO 11 ITH=ITHB,NTIP1	TMP	124
	IF (IRR(ITH).LE.NRIP1) GO TO 6	TMP	125
11	CONTINUE	TMP	126
12	CONTINUE	TMP	127
C		TMP	128
C	FRAME ADVANCE AND RETURN	TMP	129
	CALL DDFR	TMP	130
C		TMP	131
	RETURN	TMP	132

C		TMP 133
13	FORMAT (13)	TMP 134
	END	TMP 135-
	SUBROUTINE RCTOUR	RCT 1
C		RCT 2
C	SPECIAL ROUTINE TO SEARCH FOR ALL VALUES BELONGING TO A CONTOUR	RCT 3
C		RCT 4
	COMMON /DATA/ EMAGN(109,181),IRLAR(3),ICLAB(19),MODE,DELTAR,DELTAT	RCT 5
	1,NNINC,NTINC,NRIP1,NTIP1,PI,JJ(181),ESCALE	RCT 6
	COMMON /SUB/ IRK(181),IXS(181),IYS(181),PID2,PIDNTI,R,NP	RCT 7
C		RCT 8
	NP=0	RCT 9
	DO 4 ITH=1,NTIP1	RCT 10
	IRB=IRR(ITH)	RCT 11
	IF (IRB.GT.NRIP1) GO TO 4	RCT 12
	EM1=EMAGN(IRB-1,ITH)	RCT 13
	DO 1 IR=IRB,NRIP1	RCT 14
	E=EMAGN(IR,ITH)	RCT 15
	IF (EM1.LE.R.AND.R.LE.E) GO TO 2	RCT 16
	IF (EM1.GE.R.AND.R.GE.E) GO TO 2	RCT 17
	EM1=E	RCT 18
1	CONTINUE	RCT 19
	RNUM=0.0	RCT 20
	GO TO 3	RCT 21
2	RNUM=EM1-R	RCT 22
	RDEN=EM1-E	RCT 23
	RINTP=(IR-2.0+RNUM/RDEN)*DELTAR	RCT 24
	RADIUS=RINTP*450.0	RCT 25
	TT=PID2-(ITH-1)*PIDNTI	RCT 26
	NP=NP+1	RCT 27
	IXS(NP)=COS(TT)*RADIUS+512.0	RCT 28
	IYS(NP)=SIN(TT)*RADIUS+532.0	RCT 29
3	IRR(ITH)=IR+1	RCT 30
	IF (RNUM.EQ.0.0.AND.NP.NE.0) GO TO 5	RCT 31
4	CONTINUE	RCT 32
C		RCT 33
5	RETURN	RCT 34
	END	RCT 35-
	SUBROUTINE INTRPL(IU,L,X,Y,N,U,V)	0001
C	INTERPOLATION OF A SINGLE-VALUED FUNCTION	0002
C		0003
C	THIS SUBROUTINE INTERPOLATES, FROM VALUES OF THE FUNCTION	0004
C	GIVEN AS ORDINATES OF INPUT DATA POINTS IN AN X-Y PLANE	0005
C	AND FOR A GIVEN SET OF X VALUES (ABSCISSAS), THE VALUES OF	0006
C	A SINGLE-VALUED FUNCTION Y = Y(X).	0007
C		0008
C		0009
C	THE INPUT PARAMETERS ARE	0010
C		0011
C	IU = LOGICAL UNIT NUMBER OF STANDARD OUTPUT UNIT	0012
C	L = NUMBER OF INPUT DATA POINTS	0013
C	(MUST BE 2 OR GREATER)	0014
C	X = ARRAY OF DIMENSION L STORING THE X VALUES	0015
C	(ABSCISSAS) OF INPUT DATA POINTS	0016



C	(IN ASCENDING ORDER)	0017
C	Y = ARRAY OF DIMENSION L STORING THE Y VALUES	0018
C	(ORDINATES) OF INPUT DATA POINTS	0019
C	N = NUMBER OF POINTS AT WHICH INTERPOLATION OF THE	0020
C	Y VALUE (ORDINATE) IS DESIRED	0021
C	(MUST BE 1 OR GREATER)	0022
C	U = ARRAY OF DIMENSION N STORING THE X VALUES	0023
C	(ABSCISSAS) OF DESIRED POINTS	0024
C		0025
C	THE OUTPUT PARAMETER IS	0026
C		0027
C	V = ARRAY OF DIMENSION N WHERE THE INTERPOLATED Y	0028
C	VALUES (ORDINATES) ARE TO BE DISPLAYED	0029
C		0030
C		0031
C	DECLARATION STATEMENTS	0032
C		0033
	DIMENSION X(1),Y(1),U(1),V(1)	0034
	EQUIVALENCE (P0,X3),(Q0,Y3),(Q1,T3),(UK,DX),(IMN,X2,A1,FM1),	0035
	1 (IMX,X5,A5,FM5),(J,SW,SA),(Y2,W2,W4,Q2),(Y5,W3,Q3)	0036
C		0037
C	PRELIMINARY PROCESSING	0038
C		0039
	10 L0=L	0040
	LM1=L0-1	0041
	LM2=LM1-1	0042
	LP1=L0+1	0043
	N0=N	0044
	IF(LM2)90,100,100	0045
	100 IF(N0)91,91,105	0046
	105 DO 11 I=2,L0	0047
	IF(X(I-1)-X(I)) 11,95,96	0048
	11 CONTINUE	0049
	IPV=0	0050
C		0051
C	MAIN DO-LOOP	0052
C		0053
	DO 80 K=1,N0	0054
	UK=U(K)	0055
C		0056
C	ROUTINE TO LOCATE THE DESIRED POINT	0057
C		0058
	20 IF(LM2)200,27,200	0059
	200 IF(UK - X(L0))205,26,26	0060
	205 IF(UK - X(1))25,208,208	0061
	208 IMN=2	0062
	IMX=L0	0063
	21 I=(IMN+IMX)/2	0064
	IF(UK - X(I))22,23,23	0065
	22 IMX=I	0066
	GO TO 24	0067
	23 IMN=I+1	0068
	24 IF(IMX - IMN)240,240,21	0069
	240 I=IMX	0070

GO TO 30	0071
25 I=1	0072
GO TO 30	0073
26 I=LP1	0074
GO TO 30	0075
27 I=2	0076
C	0077
C CHECK IF I = IPV	0078
C	0079
30 IF(I - IPV)300,70,300	0080
300 IPV=I	0081
C	0082
C ROUTINES TO PICK UP NECESSARY X AND Y VALUES AND	0083
C TO ESTIMATE THEM IF NECESSARY	0084
C	0085
40 J=1	0086
IF(J-1)401,400,401	0087
400 J=2	0088
401 IF(J-LP1)403,402,403	0089
402 J=L0	0090
403 X3=X(J-1)	0091
Y3=Y(J-1)	0092
X4=X(J)	0093
Y4=Y(J)	0094
A3=X4-X3	0095
FM3=(Y4-Y3)/A3	0096
IF(LM2)404,43,404	0097
404 IF(J - 2)405,41,405	0098
405 X2=X(J-2)	0099
Y2=Y(J-2)	0100
A2=X3-X2	0101
FM2=(Y3-Y2)/A2	0102
IF(J - L0)41,42,41	0103
41 X5=X(J+1)	0104
Y5=Y(J+1)	0105
A4=X5-X4	0106
FM4=(Y5-Y4)/A4	0107
IF(J - 2)45,410,45	0108
410 FM2=FM3+FM3-FM4	0109
GO TO 45	0110
42 FM4=FM3+FM3-FM2	0111
GO TO 45	0112
43 FM2=FM3	0113
FM4=FM3	0114
45 IF(J - 3)46,46,450	0115
450 A1=X2-X(J-3)	0116
FM1=(Y2-Y(J-3))/A1	0117
GO TO 47	0118
46 FM1=FM2+FM2-FM3	0119
47 IF(J - LM1)470,48,48	0120
470 A5=X(J+2)-X5	0121
FM5=(Y(J+2)-Y5)/A5	0122
GO TO 50	0123
48 FM5=FM4+FM4-FM3	0124

C		0125
C	NUMERICAL DIFFERENTIATION	0126
C		0127
	50 IF (1-LP1) 500,52,500	0128
500	W2=ABS(FM4-FM3)	0129
	W3=ABS(FM2-FM1)	0130
	SW=W2+W3	0131
	IF (SW) 51,501,51	0132
501	W2=0.5	0133
	W3=0.5	0134
	SW=1.0	0135
51	T3=(W2*FM2+W3*FM3)/SW	0136
	IF (1-1) 52,54,52	0137
52	W3=ABS(FM5-FM4)	0138
	W4=ABS(FM3-FM2)	0139
	SW=W3+W4	0140
	IF (SW) 53,520,53	0141
520	W3=0.5	0142
	W4=0.5	0143
	SW=1.0	0144
53	T4=(W3*FM3+W4*FM4)/SW	0145
	IF (1-LP1) 60,530,60	0146
530	T3=T4	0147
	SA=A2+A3	0148
	T4=0.5*(FM4+FM5-A2*(A2-A3)*(FM2-FM3)/(SA*SA))	0149
	X3=X4	0150
	Y3=Y4	0151
	A3=A2	0152
	FM3=FM4	0153
	GO TO 60	0154
54	T4=T3	0155
	SA=A3+A4	0156
	T3=0.5*(FM1+FM2-A4*(A3-A4)*(FM3-FM4)/(SA*SA))	0157
	X3=X3-A4	0158
	Y3=Y3-FM2*A4	0159
	A3=A4	0160
	FM3=FM2	0161
C		0162
C	DETERMINATION OF THE COEFFICIENTS	0163
C		0164
	60 Q2=(2.0*(FM3-T3)+FM3-T4)/A3	0165
	Q3=(-FM3-FM3+T3+T4)/(A3*A3)	0166
C		0167
C	COMPUTATION OF THE POLYNOMIAL	0168
C		0169
	70 DX=UK-P0	0170
	80 V(K)=Q0+DX*(Q1+DX*(Q2+DX*Q3))	0171
	RETURN	0172
C		0173
C	ERROR EXIT	0174
C		0175
	90 WRITE (IU,2090)	0176
	GO TO 99	0177
	91 WRITE (IU,2091)	0178

GO TO 99	0179
95 WRITE (10,2095)	0180
GO TO 97	0181
96 WRITE (10,2096)	0182
97 WRITE (10,2097) 1,X(1)	0183
99 WRITE (10,2099) LD,NO	0184
RETURN	0185
C	0186
C FORMAT STATEMENTS	0187
C	0188
2090 FORMAT(1X/22H *** L = 1 OR LESS./)	0189
2091 FORMAT(1X/22H *** N = 0 OR LESS./)	0190
2095 FORMAT(1X/27H *** IDENTICAL X VALUES./)	0191
2096 FORMAT(1X/33H *** X VALUES OUT OF SEQUENCE./)	0192
2097 FORMAT(6H I =.17.10X.6HX(1) =.512.3/)	0193
2099 FORMAT(6H L =.17.10X.3HN =.17/	0194
1 36H ERROR DETECTED IN ROUTINE INTRPL/)	0195
END	0196
FUNCTION BJ(Z,N)	001
C	002
C THIS ROUTINE COMPUTES THE SPHERICAL BESSEL FUNCTIONS	003
C OF THE 1ST KIND JN(Z) FOR THE FIRST FIVE VALUES OF N	004
C	005
NP1=N+1	006
GO TO (10,20,30,40,50),NP1	007
C	008
C J0(Z)	009
10 IF(Z.NE.0.0)GO TO 11	010
BJ=1.0	011
GO TO 60	012
11 BJ=SIN(Z)/Z	013
GO TO 60	014
C	015
19 ASSIGN 30 TO JNSWT	016
GO TO 21	017
C J1(Z)	018
20 ASSIGN 60 TO JNSWT	019
21 IF(Z.NE.0.0)GO TO 22	020
BJ=0.0	021
BJ1=0.0	022
GO TO 23	023
22 BJ=(SIN(Z)-Z*COS(Z))/Z**2	024
BJ1=BJ	025
23 GO TO JNSWT,(60,30)	026
C J2(Z)	027
30 BJ=((3.0-Z**2)*SIN(Z)-3.0*Z*COS(Z))/Z**3	028
BJ2=BJ	029
IF(N-1)60,40,50	030
C	031
C J3(Z)	032
40 BJ=5.0/Z**2*BJ2-BJ1	033
GO TO 60	034
C J4(Z)	035
50 BJ=7.0/Z**3*(5.0/Z**2*BJ2-BJ1)-BJ2	036

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