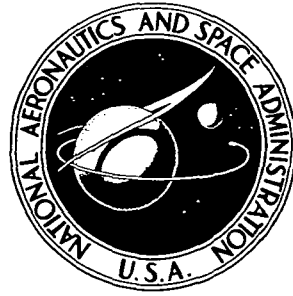


**NASA TECHNICAL  
MEMORANDUM**



**NASA TM X-3013**

**NASA TM X-3013**

**CASE FILE  
COPY**

**SOME PROPULSION SYSTEM  
NOISE DATA HANDLING CONVENTIONS  
AND COMPUTER PROGRAMS USED  
AT THE LEWIS RESEARCH CENTER**

*by Francis J. Montegani*

*Lewis Research Center*

*Cleveland, Ohio 44135*

1. Report No. <b>NASA TM X-3013</b>	2. Government Accession No.	3. Recipient's Catalog No.	
4. Title and Subtitle <b>SOME PROPULSION SYSTEM NOISE DATA HANDLING CONVENTIONS AND COMPUTER PROGRAMS USED AT THE LEWIS RESEARCH CENTER</b>		5. Report Date <b>March 1974</b>	6. Performing Organization Code
		8. Performing Organization Report No. <b>E-7609</b>	10. Work Unit No. <b>501-24</b>
7. Author(s) <b>Francis J. Montegani</b>		11. Contract or Grant No.	
9. Performing Organization Name and Address <b>Lewis Research Center National Aeronautics and Space Administration Cleveland, Ohio 44135</b>		13. Type of Report and Period Covered <b>Technical Memorandum</b>	
		14. Sponsoring Agency Code	
12. Sponsoring Agency Name and Address <b>National Aeronautics and Space Administration Washington, D.C. 20546</b>		15. Supplementary Notes	
16. Abstract <p>Methods of handling one-third-octave band noise data originating from the outdoor full-scale fan noise facility and the engine acoustic facility at the Lewis Research Center are presented. Procedures for standardizing, retrieving, extrapolating, and reporting these data are explained. Computer programs are given which are used to accomplish these and other noise data analysis tasks. This information is useful as background for interpretation of data from these facilities appearing in NASA reports and can aid data exchange by promoting standardization.</p>			
17. Key Words (Suggested by Author(s)) <b>Acoustic data analysis; Noise data analysis; Propulsion system noise; Computer programs; Data handling</b>		18. Distribution Statement <b>Unclassified - unlimited</b>  <b>Cat. 28</b>	
19. Security Classif. (of this report) <b>Unclassified</b>	20. Security Classif. (of this page) <b>Unclassified</b>	21. No. of Pages <b>70</b>	22. Price* <b>\$3.50</b>

SOME PROPULSION SYSTEM NOISE DATA HANDLING CONVENTIONS  
AND COMPUTER PROGRAMS USED AT THE  
LEWIS RESEARCH CENTER

by Francis J. Montegani

Lewis Research Center

SUMMARY

Methods of handling one-third-octave band noise data originating from the outdoor full-scale fan noise facility and the engine acoustic facility at the Lewis Research Center are presented. Procedures for standardizing, retrieving, extrapolating, and reporting these data are explained. Computer programs are given which are used to accomplish these and other noise data analysis tasks. This information is useful as background for interpretation of data from these facilities appearing in NASA reports and can aid data exchange by promoting standardization.

INTRODUCTION

The last several years have seen a rapid rise in the level of research relating to the noise of aircraft propulsion systems. Various companies, universities, and government agencies have been contributing to an increasing body of technical data. A significant volume among these data is comprised of the results of one-third-octave band analyses of the noise signatures of propulsion systems or components either from full-scale hardware or from models. Thorough analyses of these data are essential to the development of an understanding of the mechanisms of noise generation and, at the very least, to the development of noise prediction methods which are important to the proper evolution of quiet propulsion systems. Researchers using the full-scale fan and the engine acoustic facilities at the Lewis Research Center are among those generating and manipulating large volumes of one-third-octave band noise data. Several years ago, the generation of such quantities of data was anticipated, and data handling practices were established to facilitate the manipulation and analyses of these data at the user level. These practices center around a collection of computer programs which are used to process the data and

a philosophy of data standardization and storage. Extensive use of these methods has demonstrated their worth. Interest shown by others outside of NASA has indicated that these practices may be generally useful, particularly with regard to standardization and facilitation of more direct data exchange especially in computer-compatible form. In addition, considerable data from the full-scale fan facility and the engine acoustic facility are being published which are handled by the methods discussed herein. Typical of such publications is reference 1.

This report therefore sets forth the practices that are followed at the Lewis Research Center with regard to standardizing, retrieving, extrapolating, and reporting noise data from these facilities. The adjustments made to the data for standardization purposes and other practices are explained, and a collection of computer programs is given. It is hoped that these practices, which have proven useful at Lewis over the past several years, will be of use to others engaged in propulsion system noise research.

## DATA ACQUISITION AND DOCUMENTATION

This section includes information on the manner in which collections of data are identified and manipulated. To assist in presenting this information, a block diagram is presented in figure 1. This diagram gives an overview of the major elements which are involved in the data handling system. It is pertinent to much of the discussion which follows.

### Data Standardization

Measured array. - The kind of data under consideration in the context of this report are the results of one-third-octave band spectral analyses of the far-field radiated noise from axisymmetric sources. In general, measurements are made about the source at equal angle increments. Irrespective of the manner of testing, and whether one run is made or the data from a few runs are averaged, the net result is an array of data comprised of sound pressure levels in one-third-octave bands for a number of angles. This array is referred to herein as the "Measured Array" and is the starting point for all further discussion. Effects of instrumentation frequency response are presumed to be removed.

A Measured Array is identified with an operating point of the source and consists of an  $NF \times NM$  matrix of sound pressure levels, where  $NF$  is the number of frequency bands and  $NM$  is the number of microphones or angles. The usual range of frequencies is from 50 to 20 000 hertz. Microphones commonly are employed in  $10^0$  increments over most of a  $180^0$  arc.

Site effects. - The Measured Array, whether obtained indoors or out, in general possesses measurement anomalies which are attributable to the site and also to the ambient air conditions. Site-related anomalies such as ground plane reflections are the subject of continuing interest. A discussion of them exceeds the scope of this report. For full-scale fan and engine facilities, data are not adjusted routinely for site effects, but their inclusion is always implied, and it is intended that they be given consideration as the use of the data dictates.

Excess atmospheric attenuation. - Data are filed for use essentially as obtained except for standardization for atmospheric effects. Ambient air, of course, is known to cause excess sound attenuation (over and above inverse square law attenuation) which is a function of frequency, temperature, and humidity. Data to evaluate this effect are contained in reference 2.

Reference 2 was created in a framework of jet noise. And in addition to the air attenuation data contained therein, which are essentially continuous functions of frequency, guidelines are presented for applying those data to one-third-octave band spectra which are discontinuous. The guidelines specify the use of the band center frequency for determining the air attenuation for one-third-octave bands to 4000 hertz and for using the band lower limiting frequency above 4000 hertz. This procedure is biased for jet noise, which has a characteristic fall-off at high frequency. The theoretically correct attenuation which should be used must be the result of an integration which accounts for the combination of spectral and attenuation variations over a frequency band. When both these characteristics are relatively flat, the use of the attenuation at the band center frequency is appropriate. But when large changes occur in either characteristic over a frequency band, the band center frequency is not an adequate parameter. For jet noise, use of the band lower limiting frequency is satisfactory; but this is not true in general. The manner of determining atmospheric correction which is discussed in the next section uses the data of reference 2, but does so by an integration process over each frequency band. This integration cannot be done precisely with knowledge only of the one-third-octave band spectrum, but the spectrum shape is approximated conceptually by assuming a straight-line connection between sound pressure levels in adjacent frequency bands on a spectral plot with a logarithmic frequency scale.

Referred array. - The only adjustments which the Measured Array undergoes prior to use at the working level are removal of the effect of atmospheric absorption for the conditions that prevailed at the test site and adjustment to a standard radius for any microphones not on that standard radius for the test. The atmospheric attenuation for the test ambient temperature and relative humidity for each one-third-octave band spectrum is calculated as noted in the previous section. These results are added to the measured data for the appropriate propagation distance. And inverse square adjustments are made where necessary (fig. 1). The results are sound pressure levels that would exist at the microphones on a constant radius if the atmosphere were completely

nonattenuating. Therefore, these results are never to be expected in reality. It is from these data, however, that source acoustic power and directivity properties must be calculated. The array so adjusted is termed the "Referred Array," implying that it possesses acoustic properties that refer back to the source and are uninfluenced by the propagation properties of the medium except for inverse square attenuation.

It follows that a Referred Array may be extrapolated to any distance by using the inverse square law while preserving its intrinsic acoustic power and directivity properties. Conversely, when acoustic power and directivity properties of a source are known, a Referred Array can be constructed for any distance from the source (fig. 1), and by incorporating the effects of atmospheric absorption, far-field sound pressure levels may be constructed.

### Working Data

From the Referred Array, the essential properties of the source acoustic emission are calculated. These properties consist of overall power level, normalized power spectrum, and directivity index for each frequency band. These data are useful directly in the characterization of the source acoustic emission and in understanding noise generating mechanisms. Further, they are independent of the original measurement distances.

It is these data, retained on punched cards, that constitute the heart of the retrieval system. The data in this form are called "Working Data."

These noise data so decomposed into fundamental emission properties can contribute to understanding of noise generating mechanisms through the development of improved prediction techniques, for which they are suited particularly well. Each of the three basic elements - power level, normalized spectrum, and directivity index - can be examined separately and independently. Power level is a single variable which, in general, can be expected to correlate simply with size, thrust, or mechanical power. Quite independently, normalized power spectrum (which embodies only the shape and frequency scale of the spectral emission) may be expected to correlate with such things as mechanical design and characteristic speed. And finally, directivity index may be isolated and separately investigated insofar as it pertains to questions of theoretical acoustic propagation, duct terminations, flow refraction, and so forth. The extent of understanding any one of these emission properties is not dependent necessarily on the understanding of any other.

This general independence of the emission properties is also particularly useful to meet short-term needs for noise predictions. State-of-the-art noise prediction methods rely heavily on an empirical data base. Working Data facilitate such predictions by permitting selective use of the appropriate emission properties from a variety of sources.

Although Working Data, at a glance, do not consist of familiar sound pressure levels, this is no obstacle to users desirous of data in that form. With a computer-oriented system such as this, it is a simple task to assemble the Working Data into a referred sound pressure level array. This array in turn can be extrapolated for any far-field conditions (fig. 1).

For the foregoing reasons, all noise data are decomposed into the source fundamental emission properties, punched into cards, and filed for use in this form at the working level. Appendix A contains a sample listing of Working Data, with complete explanation of the format. The retention of data in this form, in conjunction with a family of computer programs which are given herein, facilitates rapid dissemination and efficient utilization of the data. The manner in which Working Data are computed from the Referred Array is discussed in the next sections.

Acoustic power. - A general sound source emits acoustic energy radially to the far-field and nonuniformly with direction. The acoustic power emitted by the source can be obtained by integrating the far-field sound intensity over an enclosing surface. It is here presumed that the source and its associated sound field are axisymmetric so that the intensity field is a function of only two coordinates, radial distance  $r$  from the source and azimuth angle  $\theta$  from the source axis. For far-field noise radiation, discounting excess atmospheric attenuation, the sound intensity varies inversely as the square of the distance from the source so that the functional dependence on  $r$  is known. Further, the sound intensity  $I$  is related to the rms acoustic pressure  $P$ , which is the usual measured quantity, according to

$$I = \frac{P^2}{\rho c} \quad (1)$$

where  $\rho c$  is the characteristic impedance of the propagating medium. Under such conditions, the rms sound pressure may be determined at a constant radius  $R$ , from which the acoustic power  $W$  is obtained according to

$$W = \frac{1}{\rho c} \int_0^\pi P^2(\theta) dA \quad (2)$$

where

$$dA = 2\pi R^2 d\theta \quad (3)$$

is the elemental annulus area on an enclosing sphere.

In practice, a function  $P^2(\theta)$  is not usually available, but rather, discrete values  $P_i^2$  ( $i = 0, 1, \dots, n$ ) are known from measurements corresponding to values of  $\theta_i$ . Under these conditions, equation (2) must be approximated as a finite sum

$$W = \frac{1}{\rho c} \sum_{i=0}^n P_i^2 \Delta A_i \quad (4)$$

where the  $\Delta A_i$  are contiguous finite incremental areas on which the corresponding  $P_i^2$  are presumed constant.

If the  $\theta_i$  are taken to be equally spaced by an angle increment  $\Delta\theta$ , which therefore becomes also the arc width of any elemental area  $\Delta A_i$ , and if the  $\theta_i$  are written as  $i\Delta\theta$ , the incremental areas can be expressed as

$$\Delta A_i = 2\pi R^2 \sin\left(\frac{\Delta\theta}{2}\right) \tan\left(\frac{\Delta\theta}{4}\right) \quad i \Delta\theta = 0, \pi \quad (5)$$

$$\Delta A_i = 4\pi R^2 \sin\left(\frac{\Delta\theta}{2}\right) \sin(i \Delta\theta) \quad 0 < i \Delta\theta < \pi \quad (6)$$

The geometry of the arrangement is shown in figure 2. When the azimuth angle  $i \Delta\theta$  corresponds with a polar area on the sphere, equation (5) applies; otherwise equation (6) applies.

Combining equations (4) to (6) results in

$$W = \frac{4\pi R^2}{\rho c} \sin\left(\frac{\Delta\theta}{2}\right) \left[ \frac{P_0^2}{2} \tan\left(\frac{\Delta\theta}{4}\right) + \sum_{i=1}^{n-1} P_i^2 \sin(i \Delta\theta) + \frac{P_n^2}{2} \tan\left(\frac{\Delta\theta}{4}\right) \right] \quad (7)$$

where the subscripts 0 and n denote the polar areas.

Equation (7) is basically that used for acoustic power calculation as discussed herein. However, as written, it applies specifically to radiation in the absence of a ground plane since a summation is taken with fully circular annuli. Most practical propulsion system noise measurements are taken in an environment with a reflecting ground plane, and the problems it presents must be considered.

In actual fact, the directly radiated instantaneous sound pressure and that reflected from the ground plane sum algebraically at the microphone. The resultant effects



depend on the geometry of the problem, the frequency of the radiation, the phase shifts and attenuations in the reflection process, and the bandwidth of the analysis technique, among other things. For one-third-octave band analysis, there usually result band-dominant signal reinforcement and cancellation effects in the low-frequency end of the spectrum which are highly dependent on the test arrangement. However, for any arrangement, many signal reinforcements and cancellations occur in any given one-third-octave band at the high-frequency end of the spectrum which, for hard reflecting surfaces, result in a theoretical doubling of intensity there. Since Lewis test sites use hard pavement reflecting surfaces for purposes of maintaining surface constancy, the doubling of intensity at high frequencies is taken to be the prevailing phenomenon. The acoustic power calculation therefore proceeds on the assumption that the acoustic intensities determined to exist in the presence of the ground plane are double what they would be in its absence. Thus, the intensities are halved for all frequency bands and summed over the entire sphere according to equation (7). No attempts are made to correct the data for ground interference effects at low frequencies.

The use of Working Data offers a convenient means of transmitting data to other users. It must be cautioned, however, that if the Referred Array is to be precisely reconstructed, the exact power calculation method which was used to generate the Working Data must be available and inverted; otherwise differences will result. The computer subroutine to compute acoustic power as given by equation (7) and the preceding discussion is called POWER and is given in appendix B. Where sound pressure level data are not available (e. g. , near the source axis, where jet flow would impinge on the microphones), the associated areas are excluded from the summation process of the power calculation.

Directivity index. - Directivity index is a normalizing way of characterizing the directional property of far-field acoustic emission. It is defined as the difference, in decibels, between the existing sound pressure level at a point and the sound pressure level that would exist at the same point from a simple source emitting the same acoustic power. Directivity index is a function of direction only; and for an axisymmetric source, therefore, it is a function of azimuth angle  $\theta$ . For discrete values of  $\theta_i$  it is defined by

$$DI_{\theta_i} = SPL_{\theta_i} - SPL_{AV} \quad (8)$$

where  $SPL_{\theta_i}$  denotes the existing sound pressure levels and  $SPL_{AV}$  represents the simple-source sound pressure level. The simple-source sound pressure level can be shown to be exactly the area-weighted average sound pressure level of the existing  $SPL_{\theta_i}$ . It is computed according to

$$\text{SPL}_{\text{AV}} = 10 \log_{10} \left( \frac{\sum_{i=0}^n \frac{\text{SPL}_{\theta_i}}{10^{10}} \cdot \Delta A_i}{\sum_{i=0}^n \Delta A_i} \right) \quad (9)$$

where the  $\Delta A_i$  are given by equations (5) and (6).

A subroutine AVSLR for computing the area-weighted sound pressure level is given in appendix B. As in the acoustic power calculation, where no data exist for angles at or near the axis, such angles are omitted from the summation process.

### Extrapolation

Often, for practical purposes, the detailed properties of the source radiation such as directivity index are not needed, but rather the far-field sound pressure levels that result when the data are extrapolated to various distances are necessary. Usually, data of this kind are reported in the literature. To generate such data, it is only necessary to assemble a Referred Array from the Working Data and extrapolate it to the desired distances by using the inverse square law and excess atmospheric attenuation for the conditions desired, usually standard day (fig. 1). Other effects such as ground reflections or extra ground attenuation are given consideration by some investigators in extrapolation calculations. However, as in the case of measurement anomalies, other extrapolation phenomena are the subject of continuing research and exceed the scope of the discussion here. These phenomena are neglected in ordinary data extrapolations for reporting purposes or for data retention at the working level. Such a practice avoids the need for qualifying the data. Further, it permits any user to quickly deduce from the data the referred arrays which accurately reflect the original data which were measured and which he may modify to suit his needs considering test site or extrapolation anomalies.

## COMPUTER PROGRAMS

### Working Data Generation

The key to the efficient retrieval and use of noise data at the working level lies in maintaining a punched card file of Working Data sets and a family of programs and sub-routines for manipulating those data. There exists, of course, archival storage of the raw measured data. But utilization of these data directly requires extensive computer interaction and program handling, particularly since the data to be so retrieved and processed consist of repeat runs which must be averaged, corrected for measurement instrumentation response when necessary, and adjusted to standard-day conditions. The use of Working Data, which is one level removed from the archival data, permits rapid data access by persons not necessarily skilled in computer usage.

The effort avoided by the routine use of Working Data in place of archival data is replaced by the one-time use of a computer program which generates the Working Data and which also generates other useful data listings. This program is called WODAG (for Working Data Generation), and an outline of the calculations it performs and the listings it generates are discussed next. WODAG is a subroutine whose principal input is a Measured Array. A main program which must provide suitable Measured Arrays and call the subroutine WODAG is the responsibility of the reader. The complete codes of the subroutine WODAG and of the subroutines which it calls are given in appendix B.

Listings. - A sample listing of the printed output generated by WODAG is presented in table I. Each page of output is somewhat self-explanatory, but they are reviewed briefly here. A summary of the pages by title is as follows:

- (1) Measured Array
- (2) Test-day excess atmospheric attenuation
- (3) Referred Array
- (4) Acoustic power computations
- (5) Normalized power spectrum (graph)
- (6) Directivity index
- (7) Atmospheric attenuation
- (8) Standard-day data excess atmospheric attenuation
- (9) Standard-day data
- (10 and following) Sideline extrapolated data

The Measured Array has been discussed. It is identifiable with a particular operating condition of the source and represents the actual measured data (or an average of data) from the test site with instrumentation frequency response characteristics removed. Variable microphone radii are permitted and appear in the listing. Atmospheric

conditions for the test are also listed. All test data as printed are adjusted to a constant radius for review purposes by using inverse square law only. For all subsequent calculations the proper atmospheric absorptions and distances are accurately accounted for.

The test-day excess atmospheric attenuation table gives the actual adjustments, based on test temperature, relative humidity, and actual measurement distances, used to generate the Referred Array.

The Referred Array is tabulated for a selected radius and, in addition, overall levels at each angle are listed. No perceived noise levels are presented since the referred data represent a condition that cannot be observed.

Results of acoustic power calculations which are listed include overall acoustic power, acoustic power spectrum, and normalized power spectrum. The normalized spectrum is obtained by subtracting the maximum band power level from the power spectrum. In addition, the simple-source sound pressure levels are listed. These are referred sound pressure levels created by a nondirectional source emitting the same acoustic power as the real source. Simple-source sound pressure levels (average sound pressure levels given by eq. (9)) are used to calculate directivity index according to equation (8). Directivity index is listed in a separate table for the sound pressure levels in each frequency band and for the overall levels.

Tables of standard-day excess atmospheric attenuation are given, the first of which is the attenuation per thousand feet of propagation distance. This table is computed from the Referred Array as discussed in the section Excess atmospheric attenuation and is used for all subsequent extrapolations. A second atmospheric attenuation table is given which lists the exact adjustments that were made to the referred array to generate the standard-day array for the same radius. The atmospheric adjustments vary from angle to angle because the spectral shapes are accounted for as previously discussed.

Subsequent tables provide data extrapolated to selected sidelines. The first of these sidelines is always at the same distance as the radius used for the standard-day data.

Punched cards. - In addition to the foregoing printed output, WODAG also punches data into cards in the Working Data format as discussed previously. These cards are intended for routine use at the user working level in conjunction with a family of programs and subroutines which are discussed next.

### General Programs and Subroutines

A principal reason for the use of Working Data is to permit convenient access to data in all its detail. Since the card data format is standardized and contains control information, one set may be read into computer storage with a simple call to a subroutine. Similar calls to other subroutines will generate Referred Arrays, do extrapo-

lations, generate perceived noise levels, and so forth. This modular approach to programming for purposes of handling the data frees the user-programmer from concern over routine data handling tasks. The use of other main programs permits nonprogrammers also to conveniently access, extrapolate, and analyze data starting with Working Data (fig. 1). A number of programs used for these various purposes are discussed in this section.

Source codes written in FORTRAN IV for all the programs which are discussed in this report and other utility subroutines necessary to support them are given in appendix B alphabetically by name. All program listings contain information which makes them self-explanatory. Many of them have a general use to those engaged in noise work. Others, described as "utility" routines, are used solely to perform very specific and mundane calling program tasks.

Following is an alphabetical summary of all the programs in appendix B with descriptions of their functions:

Program name	Description
ANGLE	(Utility)
APNDB	(Utility)
ASMBL	Assembles one set of Working Data already in storage into a Referred Array.
ATMAT	Computes excess atmospheric attenuation in decibels for any temperature, relative humidity, frequency, and distance. Uses empirical curve-fits of data contained in reference 2.
AVLSR	Computes simple-source sound pressure level (area-weighted average sound pressure level) from directional data on an arc. Results used for directivity index calculations.
BASPAT	Computes excess atmospheric attenuation for all bands of a fractional octave band spectrum considering spectrum shape. Also extrapolates spectrum to a new distance, accounting for inverse square and atmospheric attenuation.
DADIFF	Used for thorough comparison of noise characteristics of two sources. Computes the differences between two sets of data. Differences include acoustic power (including front/rear power split arbitrarily divided at 90° to the source axis), Referred Arrays, and perceived noise levels and tone-corrected perceived noise levels for standard-day data extrapolated to selected sidelines.
DBSUM	(Utility)

Program name	Description
FARDTA	Extrapolates a Referred Array to any far-field radius or sideline, accounting for inverse square and excess atmospheric attenuation.
GRAPH	(Utility)
LIST	Reads and assembles sets of Working Data and prepares printed output of the basic data and standard-day arc extrapolations including optional extrapolations to selected sideline distances. (Output identical with that of WODAG commencing with Referred Array.)
OASPL	(Utility)
PNDB	Computes perceived noise level in PNdB in accordance with reference 3.
PNLT	Computes tone-corrected perceived noise level (PNLT) for a one-third-octave band sound spectrum in accordance with reference 4. Also computes perceived noise level.
POWER	Computes total acoustic power by incremental area summation for a set of angles and referred sound pressure levels on an arc. Perfectly reflective ground plane assumed.
SIDLAT	(Utility)
TABLE	Prepares a compact one-page table of data in a format suitable for reporting purposes. The output includes standard-day extrapolated data on an arc, total acoustic power, power spectrum, simple-source sound pressure levels (which, with nominal band atmospheric attenuation, permit quick evaluation of directivity index), and optional sideline perceived noise levels.
TBLOP	(Utility)
TITLE	(Utility)
TITLE 2	(Utility)
WDATA	Reads one set of Working Data from cards into storage.
WODAG	As discussed herein, standardizes measured data, prepares data listings, and punches Working Data.

Sample output from WODAG, TABLE, and DADIFF are given herein tables I, II, and III, respectively.

## CONCLUDING REMARKS

Methods of data handling and computer programs have been presented which have proven useful for a wide variety of tasks with data from the full-scale fan and engine acoustic test facilities at the Lewis Research Center. These methods center on the use of immediately accessible data punched into cards as standard-format Working Data which include power level, normalized power spectrum, and directivity index. Working Data are useful in understanding mechanisms of generation, developing prediction methods, and executing empirical predictions. Working Data and the associated programs also simplify the problems of user-programmers and nonprogrammers in the tasks of accessing and manipulating the data and increase the productivity and quality of data analyses. It is hoped that these advantages, in addition to the information presented herein, will be of use to others and may lead ultimately to improvements in information exchange.

Lewis Research Center,  
National Aeronautics and Space Administration,  
Cleveland, Ohio, October 12, 1973,  
501-24.

## APPENDIX A

### WORKING DATA CARD FORMAT

Working Data are normalized acoustic data arranged in a standard format and used for card input for a variety of data analysis programs. A single set of data is complete and self-contained and represents the acoustic emission properties for one operating condition of a sound source.

The data are arranged in five blocks. Block 1 consists always of four cards of identifying information. This information is not manipulated in any way but is read alphanumerically. Any or all of the four cards may be blank, and all 80 card columns may be used.

Block 2 consists of a single card providing control data and identification of the particular operating conditions for this data set. The data, location, and format on the block 2 card are as follows:

Card column	1	5	13	21	24	27
Variable	NCONF	RPM	PCS	NF	NM	NB
FORTTRAN format	I4	F8.1	F8.1	I3	I3	I3

NCONF configuration number

RPM speed, rpm

PCS percent speed

NF number of fractional-octave band filters employed, up to 27

NM number of equally spaced angles for the data array, up to 19

NB 1/NB -octave frequency bands

Obviously, where NCONF, RPM, and PCS as defined are inappropriate for the sound source, other operating variables may be used.

Block 3 consists of one, two, or three cards, depending on the value of NF. The first card is arranged in the following way:

Card column	1	7	13	19	...	67
Variable name	PWL	SUMN	PSM(1)	PSM(2)	...	PSM(10)
FORTTRAN format	F6.1	F6.1	F6.1	F6.1	...	F6.1



PWL total acoustic power level, dB referenced to  $10^{-13}$  W  
 SUMN antilogarithmic sum in decibels of the normalized power spectrum  
 PSM(I) normalized power spectrum, the power spectrum from which has been subtracted the maximum band level

Cards 2 and 3 of block 3, when they exist, are arranged as follows:

Card column	1	13	19	...	67
Variable	(Blank)	PSM(11)	PSM(12)	...	PSM(20)
FORTTRAN format	12x	F6.1	F6.1	...	F6.1

Card column	1	13	19	...	49
Variable	(Blank)	PSM(21)	PSM(22)	...	PSM(27)
FORTTRAN format	12x	F6.1	F6.1	...	F6.1

The index of PSM(I) terminates with the value of NF.

Block 4 consists of at least one card having the increment angle DT followed by the actual microphone angles AI(J), continuing on to the first column of a second card if necessary. The format is 12F6.1/8F6.1. Not counting the increment angle, the number of angles agrees with the value of NM.

Block 5 cards contain the directivity index DI(I, J) for each frequency band denoted by both a band number I and the band center frequency NFIL(I). The subscript J denotes angle. Block 5 consists of NF sets of one or two cards each. The card formats for the I<sup>th</sup> set are

Card column	1	7	13	19	...	67
Variable name	I	NFIL(I)	DI(I, 1)	DI(I, 2)	...	DI(I, 10)
FORTTRAN format	I6	I6	F6.1	F6.1	...	F6.1

Card column	1	13	...	61	...
Variable name	(Blank)	DI(I, 11)	...	DI(I, 19)	...
FORTTRAN format	12x	F6.1	...	F6.1	...

The index  $J$  of  $DI(I, J)$  terminates with the value of  $NM$ .

A listing of a typical set of working data is given in table IV. The maximum number of cards in a set is 64.



```

SUBROUTINE ASMBL (R,SL)
C          /ASMBL - ASSEMBLE DATA/
C * * * * *
C *
C * ASSEMBLES ONE SET OF WORKING DATA FROM COMMON BLOCK /WD/ INTO
C * A REFERRED ARRAY.
C *
C * R          RADIUS FOR REFERRED ARRAY
C * SL        REFERRED ARRAY
C *
C * * * * *
C * DIMENSION SL(27,19)
C * COMMON /WD/A(20,4),NCONF,RPM,PCS,NF,NM,NB,PWL,SUMN,PSM(27),DT,
C * 1 AI(19),NFIL(27),DI(27,19)
C * * * * *
C *
C * VARIABLES IN COMMON BLOCK /WD/
C *
C * A(20,4)   FOUR CARDS OF ID (WORD LENGTH 4)
C * NCONF     CONFIGURATION NUMBER
C * RPM       SPEED IN RPM
C * PCS       PERCENT SPEED
C * NF        NUMBER OF FREQUENCY BANDS
C * NM        NUMBER OF ANGLES
C * NB        1/NB-OCTAVE BANDS
C * PWL       OVERALL ACOUSTIC POWER LEVEL
C * SUMN      DECIBEL SUM OF NORMALIZED POWER SPECTRUM
C * PSM(27)   NORMALIZED POWER SPECTRUM
C * DT        ANGLE INCREMENT
C * AI(19)    ANGLES
C * NFIL(27)  BAND CENTER FREQUENCIES
C * DI(27,19) DIRECTIVITY INDEX
C *
C * * * * *
C * F=3.1415927/180.0
C * RHO=0.0023769
C * C=1116.3975
C * CONST=2.0*59.141053*1.0E-15
C * SPHERE=4.0*3.1415927*R**2
C * NFT=1
C * NLT=NM
C * SUM=0.0
C * IF (AI(1).GT.0.0) GO TO 1
C * SUM=SUM+1.0
C * NFT=2
1  IF (AI(NM).LT.130.0) GO TO 2
C * SUM=SUM+1.0
C * NLT=NM-1
2  SUM=TAN(DT/4.0*F)/2.0*SUM
C * DO 3 J=NFT,NLT
3  SUM=SUM+SIN(AI(J)*F)
C * FAC=CONST/(RHO*C)*SPHERE*SIN(DT/2.0*F)*SUM
C * DELTA=130.0+10.0*ALOG10(FAC)
C * PPSUM=PWL-SUMN-DELTA
C * DO 4 I=1,NF
C * PSUM=PPSUM+PSM(I)
C * DO 4 J=1,NM
4  SL(I,J)=PSUM+DI(I,J)
C * RETURN
C * END

```

```

SUBROUTINE ATMAT (T,RH,DIST,FREQ,ATT)
C /ATMAT - ATMOSPHERIC ATTENUATION/
C * * * * *
C *
C * COMPUTES EXCESS ATMOSPHERIC ATTENUATION IN DECIBELS FOR GIVEN
C * TEMPERATURE, RELATIVE HUMIDITY, DISTANCE, AND FREQUENCY.
C * USES EMPIRICAL CURVE-FITS OF DATA CONTAINED IN SOCIETY OF
C * AUTOMOTIVE ENGINEERS AEROSPACE RECOMMENDED PRACTICE NO. 866,
C * AUGUST, 1964.
C *
C * T TEMPERATURE (DEGREES FAHRENHEIT)
C * RH RELATIVE HUMIDITY
C * DIST DISTANCE (FEET)
C * FREQ FREQUENCY (HERTZ)
C * ATT ATTENUATION (DECIBELS)
C *
C * * * * *
DIMENSION A(22)
DATA A/0.870,0.750,0.652,0.570,0.505,0.452,0.406,0.369,0.335,
1 0.308,0.286,0.268,0.253,0.240,0.231,0.225,0.220,0.215,0.210,
2 0.208,0.202,0.200/
AC=(0.1*(FREQ/1000.0)**2.05)/(1.651-.00103*T)**2.05
AMM=(10.0*(FREQ/1000.0)**1.003)/10.0**(0.52-.00504*(T+SQRT(256.0-(
110.0-T/5.0)**2)))
HA=0.25*RH/10.0**(1.493-.01638*T-.02*SQRT(128.2-(10.0-T/5.0)**2))
HMM=10.0**(0.4973*ALOG10(FREQ)-1.4894)
HH=HA/HMM
IF (HH.GT.0.25) GO TO 1
AA=1.2*HH
GO TO 8
1 IF (HH.GT.0.60) GO TO 2
AA=1.543*HH-.086
GO TO 8
2 IF (HH.GT.0.95) GO TO 3
AA=0.84+0.16*SIN(3.14159/2.0*(HH-0.6)/0.35)
GO TO 8
3 IF (HH.GT.1.25) GO TO 4
AA=0.87+0.13*COS(3.14159/2.0*(HH-0.95)/0.3)
GO TO 8
4 IF (HH.GT.6.5) GO TO 7
HTEST=1.25
DO 5 I=2,22
HTEST=HTEST+0.25
IF (HH.LE.HTEST) GO TO 6
5 CONTINUE
6 AA=A(I)+((HTEST-HH)/0.25)*(A(I-1)-A(I))
GO TO 8
7 AA=0.2
8 CONTINUE
ATT=(AMM*AA+AC)*(DIST/1000.0)
RETURN
END

```

```

SUBROUTINE AVSLR (SL, AI, DT, NM, SLR)
/AVSLR - AVERAGE SL ON AN ARC/
* * * * *
C * * * * *
C * COMPUTES SIMPLE-SOURCE SOUND PRESSURE LEVEL (AREA-WEIGHTED
C * AVERAGE SOUND PRESSURE LEVEL) FROM DIRECTIONAL DATA ON AN ARC.
C * RESULTS USED FOR DIRECTIVITY INDEX CALCULATION.
C *
C * SL      REFERRED DATA ON AN ARC
C * AI      CORRESPONDING ANGLES
C * DT      ANGLE INCREMENT
C * NM      NUMBER OF ANGLES
C * SLR     AVERAGE SOUND PRESSURE LEVEL
C *
C * * * * *
DIMENSION SL(19), AI(19)
F=3.1415927/180.0
NFT=1
NLT=NM
SUM=0.0
SUMD=0.0
IF (AI(1).GT.0.0) GO TO 1
SUM=SUM+10.0**(SL(1)/10.0)
SUMD=SUMD+1.0
NFT=2
1 IF (AI(NM).LT.180.0) GO TO 2
SUM=SUM+10.0**(SL(NM)/10.0)
SUMD=SUMD+1.0
NLT=NM-1
2 SUM=TAN(DT/4.0*F)/2.0*SUM
SUMD=TAN(DT/4.0*F)/2.0*SUMD
DO 3 J=NFT, NLT
SUM=SUM+10.0**(SL(J)/10.0)*SIN(AI(J)*F)
SUMD=SUMD+SIN(AI(J)*F)
3 SLR=10.0*ALOG10(SUM/SUMD)
RETURN
END

```

```

SUBROUTINE BASPAT (A,RT,NF,NB,T,RH,TFA,R,ATA)
C /BASPAT - BAND SPECTRUM ATTENUATION/
C * * * * *
C * COMPUTES FRACTIONAL-OCTAVE BAND EXCESS ATMOSPHERIC ATTENUATION
C * CONSIDERING SPECTRUM SHAPE.  EXTRAPOLATES SPECTRUM TO NEW
C * DISTANCE ACCOUNTING FOR INVERSE SQUARE AND EXCESS ATMOSPHERIC
C * ATTENUATION.
C *
C * A      REFERRED SPECTRUM
C * RT     CORRESPONDING RADIUS
C * NF     NUMBER OF FREQUENCY BANDS
C * NB     1/NB-OCTAVE FREQUENCY BANDS
C * T      TEMPERATURE (DEGREES FAHRENHEIT)
C * RH     RELATIVE HUMIDITY
C * TFA    ATTENUATION IN DECIBELS PER THOUSAND FEET
C * R      RADIUS FOR ATTENUATED SPECTRUM
C * ATA    ATTENUATED SPECTRUM AT R
C *        (INVERSE SQUARE AND ATMOSPHERIC ATTENUATION)
C *
C * CALLS  ATMAT
C *
C * * * * *
DIMENSION A(27),TFA(27),ATA(27)
REAL M(27,2)
NLS=NF-1
DO 1 I=1,NLS
M(I,2)=A(I+1)-A(I)
1 M(I+1,1)=M(I,2)
M(1,1)=M(1,2)
M(NF,2)=M(NF,1)
NS=3
S=NS
B=NB
BI=B*2.0*FLOAT(NS)
F1=1.0
IF (NB.EQ.1) F1=10.0**1.8
IF (NB.EQ.3) F1=10.0**1.7
DO 3 I=1,NF
FC=10.0**(0.3/B*FLOAT(I-1))*F1
CALL ATMAT (T,RH,1000.0,FC,AC)
SL=1.0
SLA=10.0**(-AC/10.0)
DO 2 K=1,NS
XL=-K
XR=K
EL=M(1,1)/2.0*XL/S
ER=M(1,2)/2.0*XR/S
SL=SL+10.0**((EL/10.0)+10.0**((ER/10.0)
FL=10.0**(0.3/BI*XL)*FC
FR=10.0**(0.3/BI*XR)*FC
CALL ATMAT (T,RH,1000.0,FL,AL)
CALL ATMAT (T,RH,1000.0,FR,AR)
2 SLA=SLA+10.0**(((EL-AL)/10.0)+10.0**(((ER-AR)/10.0)
TFA(I)=10.0*ALOG10(SL)-10.0*ALOG10(SLA)
3 ATA(I)=A(I)-TFA(I)*R/1000.0-20.0*ALOG10(R/RT)
RETURN
END

```

```

C MAIN PROGRAM DADIFF 1
C /DADIFF - DATA DIFFERENCE/ 2
C * * * * * 3
C * 4
C * COMPUTES DIFFERENCES BETWEEN TWO SETS OF DATA, INCLUDING 5
C * ACOUSTIC POWER, REFERRED ARRAY, AND EXTRAPOLATED DATA. 6
C * 7
C * CALLS ASMBL, ATMAT, BASPAT, DBSUM, PNDB, PNLT, 8
C * POWER, TITLE2, WDATA 9
C * 10
C * * * * * 11
COMMON A(20,4,2),AI(19,2),NFIL(27,2),NCONF(2),RPM(2),PCS(2), 12
1 NB(2),NF(2),NM(2),DT(2) 13
COMMON /WD/AA(20,4),NCON,KP,PC,NNF,NNM,NNB,PW,SUMN,PS(27),DTT, 14
1AAI(19),NFI(27),DII(27,19) 15
DIMENSION DI(27,19,2),TSL(27,19,2),TBL(27,19,2) 16
DIMENSION PSM(27,2),AF(19,2),AR(19,2),F(19,2),R(19,2),TW(27,2), 17
1 TPWL(27,2),FPSM(27,2),RPSM(27,2),DIF(27,19) 18
DIMENSION SUM(19,2),FW(27,2),RW(27,2) 19
DIMENSION PWL(2),SOM(2),WF(27),W(27),WR(27),PWF(2),PWR(2), 20
1 DELTF(27),DELTR(27),DIFF(29) 21
DIMENSION PNL(19,2),DPNL(19),S(5),TFA(27),ATA(27,19,2),PNT(19,2), 22
1DPNT(19) 23
LOGICAL FTEST,ATEST,SPLIT(2) 24
THREE=10.0*ALOG10(2.0) 25
T=59.0 26
RH=70.0 27
WRITE (6,1) 28
FORMAT (1H1) 29
C 30
READ (5,2) (S(I),I=1,5) 31
FORMAT (5F6.0) 32
C * * * * * 33
C * 34
C * INPUT DATA REQUIRED 35
C * 36
C * ONE CARD WITH UP TO FIVE SIDELINE DISTANCES, 37
C * OR A BLANK CARD FOR NO SIDELINE EXTRAPOLATIONS. 38
C * 39
C * * * * * 40
DO 3 I=1,5 41
IF (S(I).LE.0.0) GO TO 4 42
3 CONTINUE 43
NS=5 44
GO TO 5 45
4 NS=I-1 46
5 CONTINUE 47
6 CONTINUE 48
C 49
C READ, ASSEMBLE TWO SETS OF DATA 50
C 51
C DO 10 K=1,2 52
C 53
C CALL WDATA 54
C * * * * * 55
C * 56
C * INPUT DATA REQUIRED 57
C * 58
C * ONE OR MORE PAIR OF SETS OF WORKING DATA. 59
C * DIFFERENCES TAKEN USING SET TWO MINUS SET ONE. 60
C * 61
C * * * * * 62

```



	DO 7 J=1,4	63
	DO 7 I=1,20	64
7	A(I,J,K)=AA(I,J)	65
	NCCNF(K)=NCCN	66
	RPM(K)=RP	67
	PCS(K)=PC	68
	NF(K)=NNF	69
	NM(K)=NNM	70
	NB(K)=NNB	71
	PWL(K)=PW	72
	DT(K)=DTT	73
	DO 8 J=1,NNM	74
8	AI(J,K)=AAI(J)	75
	DO 9 I=1,NNF	76
	PSM(I,K)=PS(I)	77
	NFIL(I,K)=NFI(I)	78
	DO 9 J=1,NNM	79
9	DI(I,J,K)=DII(I,J)	80
	CALL ASMBL (100.0,TSL(1,1,K))	81
10	CONTINUE	82
	FTEST=.TRUE.	83
	ATEST=.TRUE.	84
	SPLIT(1)=.FALSE.	85
	SPLIT(2)=.FALSE.	86
	DO 22 K=1,2	87
	NMM=NMM(K)	88
	DO 11 J=1,NMM	89
	IF (AI(J,K).EQ.90.0) GO TO 12	90
11	CONTINUE	91
	GO TO 13	92
12	SPLIT(K)=.TRUE.	93
13	CONTINUE	94
	JS=J	95
	NFF=NF(K)	96
	DO 14 I=1,NFF	97
	DO 14 J=1,NMM	98
14	TBL(I,J,K)=TSL(I,J,K)	99
	DO 15 I=1,NFF	100
15	TSL(I,JS,K)=TSL(I,JS,K)-THREE	101
	DO 16 J=1,JS	102
16	AF(J,K)=AI(J,K)	103
	DO 17 J=JS,NMM	104
	I=J-JS+1	105
17	AR(I,K)=AI(J,K)	106
	NMF=JS	107
	NMR=NMM-JS+1	108
	DO 20 I=1,NFF	109
	DO 18 J=1,JS	110
18	F(J,K)=TSL(I,J,K)	111
	DO 19 J=JS,NMM	112
	JJ=J-JS+1	113
19	R(JJ,K)=TSL(I,J,K)	114
	CALL POWER (F(1,K),100.0,AF(1,K),NMF,DT(K),59.0,29.92,FPSM(I,K),FW	115
	1(I,K))	116
	CALL POWER (R(1,K),100.0,AR(1,K),NMR,DT(K),59.0,29.92,RPSM(I,K),RW	117
	1(I,K))	118
	TW(I,K)=FW(I,K)+RW(I,K)	119
	TPWL(I,K)=130.0+10.0*ALOG10(TW(I,K))	120
20	CONTINUE	121
	WF(K)=0.0	122
	W(K)=0.0	123
	WR(K)=0.0	124

	DO 21 I=1,NFF	125
	WF(K)=WF(K)+FW(I,K)	126
	W(K)=W(K)+TW(I,K)	127
21	WR(K)=WR(K)+RW(I,K)	128
	PWF(K)=130.0+10.0*ALOG10(WF(K))	129
	PWL(K)=130.0+10.0*ALOG10(W(K))	130
	PWR(K)=130.0+10.0*ALOG10(WR(K))	131
22	CONTINUE	132
	DPWL=PWL(2)-PWL(1)	133
	NFF=MIN0(NF(1),NF(2))	134
	DO 23 I=1,NFF	135
	IF (NFIL(I,1).NE.NFIL(I,2)) FTEST=.FALSE.	136
23	CONTINUE	137
C		138
C	SHIFT SPECTRUM ARRAYS TO GET FREQUENCY CORRESPONDENCE	139
C		140
	IF (NB(1).EQ.NB(2).AND..NOT.FTEST) GO TO 24	141
	GO TO 30	142
24	IF (NFIL(1,2).GT.NFIL(1,1)) GO TO 25	143
	LL=2	144
	LH=1	145
	GO TO 26	146
25	LL=1	147
	LH=2	148
26	CONTINUE	149
	II=NF(LL)	150
	DO 27 I=1,II	151
	IF (NFIL(I,LL).EQ.NFIL(1,1)) GO TO 28	152
27	CONTINUE	153
28	ID=I	154
	NF(LL)=NF(LL)-ID+1	155
	II=NF(LL)	156
	DO 29 I=1,II	157
	K=I+ID-1	158
	NFIL(I,LL)=NFIL(K,LL)	159
	JJ=NM(LL)	160
	DO 29 J=1,JJ	161
29	TBL(I,J,LL)=TBL(K,J,LL)	162
	NFF=MIN0(NF(1),NF(2))	163
	FTEST=.TRUE.	164
C		165
C	SHIFT ANGLE ARRAYS TO GET ANGLE CORRESPONDENCE	166
C		167
30	CONTINUE	168
	NMM=MIN0(NM(1),NM(2))	169
	IF (DT(1).EQ.DT(2)) GO TO 31	170
	ATEST=.FALSE.	171
	GO TO 37	172
31	IF (AI(1,2).GT.AI(1,1)) GO TO 32	173
	LL=2	174
	LH=1	175
	GO TO 33	176
32	LL=1	177
	LH=2	178
33	CONTINUE	179
	JJ=NM(LL)	180
	DO 34 J=1,JJ	181
	IF (AI(J,LL).EQ.AI(1,LH)) GO TO 35	182
34	CONTINUE	183
	GO TO 37	184
35	JD=J	185
	NM(LL)=NM(LL)-JD+1	186

	JJ=NM(LL)	187
	DO 36 J=1,JJ	188
	K=J+JU-1	189
	AI(J,LL)=AI(K,LL)	190
	II=NF(LL)	191
	DO 36 I=1,II	192
36	TBL(I,J,LL)=TBL(I,K,LL)	193
	NMM=MINO(NM(1),NM(2))	194
	ATEST=.TRUE.	195
C		196
C	PAGE ONE OUTPUT	197
C		198
37	CONTINUE	199
	CALL TITLE2	200
	WRITE (6,38)	201
38	FORMAT (1H ,40X,45HP D W E R L E V E L D I F F E R E N C E S//	202
	147X,33H(DATA SET TWO MINUS DATA SET ONE)//)	203
	IF (FTEST.AND.SPLIT(1).AND.SPLIT(2)) GO TO 45	204
	WRITE (6,39) PWL(2),PWL(1),DPWL	205
39	FORMAT (1H ,22X,11HTOTAL POWER//15X,27HSET TWO SET ONE DELTA PWL	206
	1//4X,8HOVERALL ,3F9.1)	207
	IF (FTEST) GO TO 41	208
	WRITE (6,40)	209
40	FORMAT (1H2,30X,83H(FREQUENCIES INCOMPATIBLE FOR COMPARISON, NO 90	210
	1 DEGREE ANGLE TO PERMIT 90/90 SPLIT))	211
	GO TO 54	212
41	WRITE (6,42)	213
42	FORMAT (1H ,3X,8HOVERALL ,3F9.1//15H BAND FREQUENCY)	214
	DO 43 I=1,NFF	215
	DELTA=TPWL(I,2)-TPWL(I,1)	216
43	WRITE (6,44) I,NFIL(I,1),TPWL(I,2),TPWL(I,1),DELTA	217
44	FORMAT (1H ,I3,I8,3F9.1)	218
	GO TO 54	219
45	WRITE (6,46)	220
46	FORMAT (1H ,22X,11HTOTAL POWER,45X,14HFRONT QUADRANT,19X,13HREAR Q	221
	1UADRANT//15X,27HSET TWO SET ONE DELTA PWL,26X,2(5X,27HSET TWO S	222
	2ET ONE DELTA PWL)//)	223
	DPF=PWF(2)-PWF(1)	224
	DPR=PWR(2)-PWR(1)	225
	WRITE (6,47) PWL(2),PWL(1),DPWL,PWF(2),PWF(1),DPF,PWR(2),PWR(1),DP	226
	1R	227
47	FORMAT (1H ,3X,8HOVERALL ,3F9.1,24X,8HOVERALL ,3F9.1,5X,3F9.1//)	228
	IF (FTEST) GO TO 49	229
	WRITE (6,48)	230
48	FORMAT (1H2,41H(FREQUENCIES INCOMPATIBLE FOR COMPARISON))	231
	GO TO 54	232
49	WRITE (6,50)	233
50	FORMAT (1H ,15H BAND FREQUENCY,45X,14HBAND FREQUENCY)	234
	DO 51 I=1,NFF	235
	DELTF(I)=FPSM(I,2)-FPSM(I,1)	236
51	DELTR(I)=RPSM(I,2)-RPSM(I,1)	237
	DO 52 I=1,NFF	238
	DELTA=TPWL(I,2)-TPWL(I,1)	239
52	WRITE (6,53) I,NFIL(I,1),TPWL(I,2),TPWL(I,1),DELTA,I,NFIL(I,1),FPS	240
	1M(I,2),FPSM(I,1),DELTF(I),RPSM(I,2),RPSM(I,1),DELTR(I)	241
53	FORMAT (1H ,I3,I8,3F9.1,21X,I3,I8,3F9.1,5X,3F9.1)	242
C		243
C	PAGE TWO OUTPUT	244
C		245
54	WRITE (6,1)	246
	CALL TITLE2	247
	WRITE (6,55)	248
55	FORMAT (1H ,32X,61HD I F F E R E N C E S O F R E F E R R E D	249

	I D A T A //47X,33H(DATA SET TWO MINUS DATA SET ONE)//)	250
	IF (ATEST) GO TO 57	251
	WRITE (6,56)	252
56	FORMAT (1H2,45H(DATA SET ANGLES INCOMPATIBLE FOR COMPARISON))	253
	GO TO 6	254
57	WRITE (6,58) (AI(J,1),J=1,NMM)	255
58	FORMAT (1H ,9X,5HANGLE,19F6.0)	256
	WRITE (6,59)	257
59	FORMAT (//)	258
	DO 60 K=1,2	259
	DO 60 J=1,NMM	260
60	CALL DBSUM (TBL(1,J,K),NFF,SUM(J,K))	261
	DO 61 J=1,NMM	262
61	DIFF(J)=SUM(J,2)-SUM(J,1)	263
	WRITE (6,62) (DIFF(J),J=1,NMM)	264
62	FORMAT (1H ,7X,7HVERALL,19F6.1)	265
	IF (FTEST) GO TO 64	266
	WRITE (6,63)	267
63	FORMAT (1H2,30X,50H(DATA SET FREQUENCIES INCOMPATIBLE FOR COMPARI ION))	268
	GO TO 69	269
64	DO 65 I=1,NFF	270
	DO 65 J=1,NMM	271
65	DIF(I,J)=TBL(I,J,2)-TBL(I,J,1)	272
	WRITE (6,66)	273
66	FORMAT (1H ,14HBAND FREQUENCY)	274
	DO 67 I=1,NFF	275
67	WRITE (6,68) I,NFIL(I,1),(DIF(I,J),J=1,NMM)	276
68	FORMAT (1H ,13,18,3X,19F6.1)	277
69	CONTINUE	278
C		279
C	PAGE THREE OUTPUT	280
C		281
	IF (NS.LE.0) GO TO 6	282
	WRITE (6,1)	283
	CALL TITLE2	284
	WRITE (6,70) T,RH	285
70	FORMAT (1H ,20X,83HPERCEIVED AND TONE-CORRECTED RECEIVED NOISE LE VELS AND DIFFERENCES ALONG SIDELINES///50X,F5.1,3H F,,F5.1,11H PER 2CENT RH)	286
	DO 81 KK=1,NS	287
	IF (KK.EQ.3.OR.KK.EQ.5) WRITE (6,1)	288
	IF (KK.EQ.3.OR.KK.EQ.5) CALL TITLE2	289
	IF (KK.EQ.3.OR.KK.EQ.5) WRITE (6,70) T,RH	290
	DO 71 K=1,2	291
	NM1=1	292
	IF (AI(1,K).LE.0.0) NM1=2	293
	IF (AI(NMM,K).GE.180.0) NMM=NMM-1	294
	DO 71 J=NM1,NMM	295
	RDIST=S(KK)/SIN(AI(J,K)*3.1415927/180.0)	296
	CALL BASPAT (TBL(1,J,K),100.0,NFF,NB,T,RH,TFA,RDIST,ATA(1,J,K))	297
71	CALL PNL (ATA(1,J,K),PNL(J,K),PNT(J,K))	298
	DO 72 J=NM1,NMM	299
72	DPNL(J)=PNL(J,2)-PNL(J,1)	300
	WRITE (6,73) S(KK),(AI(J,1),J=NM1,NMM)	301
73	FORMAT (///,11X,F6.0,12H FT SIDELINE//5X,5HANGLE,5X,19F6.0)	302
	WRITE (6,74) (PNL(J,2),J=NM1,NMM)	303
74	FORMAT (1H0,1X,12HSET TWO PNDB,1X,19F6.1)	304
	WRITE (6,75) (PNL(J,1),J=NM1,NMM)	305
75	FORMAT (1H ,1X,12HSET ONE PNDB,1X,19F6.1)	306
	WRITE (6,76) (DPNL(J),J=NM1,NMM)	307
76	FORMAT (1H0,1X,10HDELTA PNDB,3X,19F6.1)	308
	DO 77 J=NM1,NMM	309
		310
		311
		312

```

77 DPNT(J)=PNT(J,2)-PNT(J,1) 313
   WRITE (6,78) (PNT(J,2),J=NMI,NMM) 314
78 FORMAT (1H0,1X,12HSET TWO PNLT,1X,19F6.1) 315
   WRITE (6,79) (PNT(J,1),J=NMI,NMM) 316
79 FGRMAT (1H ,1X,12HSET ONE PNLT,1X,19F6.1) 317
   WRITE (6,80) (DPNT(J),J=NMI,NMM) 318
80 FORMAT (1H0,1X,10HDELTA PNLT,3X,19F6.1) 319
81 CONTINUE 320
   WRITE (6,1) 321
   GO TO 6 322
   END 323-

```

```

SUBROUTINE DBSUM (A,N,SUM) 1
C /DBSUM - DECIBEL SUM/ 2
C * * * * * 3
C * 4
C * UTILITY ROUTINE TO COMPUTE A DECIBEL SUM (ANTILOGARITHMIC SUM) * 5
C * FOR A NUMBER OF LEVELS. * 6
C * 7
C * A ARRAY OF DECIBEL VALUES * 8
C * N NUMBER OF VALUES * 9
C * SUM DECIBEL SUM * 10
C * 11
C * * * * * 12
   DIMENSION A(27) 13
   SUM=0.0 14
   DO 1 I=1,N 15
1 SUM=SUM+10.0**(A(I)/10.0) 16
   SUM=10.0*ALOG10(SUM) 17
   RETURN 18
   END 19-

```

```

SUBROUTINE FARDTA (SL,RT,NF,NB,NM,AI,T,RH,DIST,IC,FARSL)
/FA*DTA - FAR DATA/
* * * * *
*
* EXTRAPOLATES REFERRED ARRAY TO A FAR FIELD RADIUS OR SIDELINE
* ACCOUNTING FOR INVERSE SQUARE AND EXCESS ATMOSPHERIC ATTEN-
* UATION.
*
* SL      REFERRED ARRAY
* RT      CORRESPONDING RADIUS
* NF      NUMBER OF FREQUENCY BANDS
* NB      1/NB-OCTAVE FREQUENCY BANDS
* NM      NUMBER OF ANGLES
* AI      ANGLES
* T       TEMPERATURE (DEGREES FAHRENHEIT)
* RH      RELATIVE HUMIDITY
* DIST    DISTANCE FROM SOURCE FOR EXTRAPOLATION
* IC      CONTROL. IF IC EQUALS ZERO, DIST IS TAKEN AS A
*         RADIUS ABOUT THE SOURCE. IF IC = 1, DIST IS TAKEN
*         AS NORMAL DISTANCE TO A PARALLEL SIDELINE.
* FARSL   EXTRAPOLATED DATA ARRAY ON A RADIUS OR SIDELINE
*
* CALLS   BASPAT
*
* * * * *
DIMENSION SL(27,19),AI(19),FARSL(27,19),TFA(27)
RDIST=DIST
F=3.1415927/180.0
DO 6 J=1,NM
IF (IC.EQ.0) GO TO 3
IF (IC.EQ.1) GO TO 2
WRITE (6,1)
1 FORMAT (62H SIDELINE OR RADIUS NOT CORRECTLY SPECIFIED, SUBROUTINE
1 FARDTA).
RETURN
2 ST=SIN(AI(J)*F)
IF (ST.LE.0.0) GO TO 4
RDIST=DIST/ST
3 CALL BASPAT (SL(1,J),RT,NF,NB,T,RH,TFA,RDIST,FARSL(1,J))
GO TO 6
4 DO 5 I=1,NF
FARSL(I,J)=0.0
5 CONTINUE
6 RETURN
END

```

```

SUBROUTINE GRAPH (SL,NF)
/GR*APH - GRAPH OUTPUT ON PRINTER/
* * * * *
*
* UTILITY ROUTINE TO PRINT GRAPH OF A NORMALIZED ARRAY.
*
* SL      NORMALIZED ARRAY
* NF      NUMBER OF ELEMENTS OF ARRAY
*
* * * * *
DIMENSION SL(27),P(132),D(4)
DATA BLANK/1H /

```

	DATA X,PI,ZERO, SIGN,D/1HG,1HI,1H0,1H-,1HI,1H2,1H3,1H4/	13
	WRITE (6,1)	14
1	FORMAT (50X,34HNORMALIZED POWER SPECTRUM        ///)	15
	DO 2 I=1,NF	16
2	SL(I)=ABS(SL(I))	17
	DO 9 L=1,41	18
	DO 3 J=1,132	19
3	P(J)=BLANK	20
	P(24)=PI	21
	P(4*NF+20)=PI	22
	IF (L.EQ.1) P(21)=ZERO	23
	IF (L.EQ.11) GO TO 4	24
	IF (L.EQ.21) GO TO 4	25
	IF (L.EQ.31) GO TO 4	26
	IF (L.EQ.41) GO TO 4	27
	GO TO 7	28
4	P(20)=SIGN	29
	P(22)=ZERO	30
	K=L/10	31
	P(21)=D(K)	32
	IF (L.EQ.41) GO TO 5	33
	GO TO 7	34
5	KK=4*NF+19	35
	DO 6 K=25,KK	36
6	P(K)=SIGN	37
7	A=FLOAT(L-1)-0.5	38
	B=FLOAT(L-1)+0.5	39
	DO 8 I=1,NF	40
	IF (SL(I).GE.A.AND.SL(I).LT.B) P(4*I+20)=X	41
8	CONTINUE	42
9	WRITE (6,10) (P(I),I=1,132)	43
10	FORMAT (1H,132A1)	44
	WRITE (6,11) (I,I=1,NF)	45
11	FORMAT (/21X,27I4)	46
	RETURN	47
	END	48-

C	MAIN PROGRAM LIST	1
C	/ LIST - LIST DATA ON PRINTER/	2
C	* * * * *	3
C		4
C	* READS, ASSEMBLES SETS OF WORKING DATA AND PREPARES PRINTED	5
C	* OUTPUT IDENTICAL WITH THAT OF WODAG COMMENCING WITH REFERRED	6
C	* ARRAY.	7
C	*	8
C	* * * * *	9
	COMMON /WD/A(20,4),NCONF,RPM,PCS,NF,NM,NB,PWL,SUMN,PSM(27),DT,	10
	1 AI(19),NFIL(27),DI(27,19)	11

```

C * * * * * 12
C * 13
C * VARIABLES IN COMMON BLOCK /WD/ 14
C * 15
C * A(25,4) FOUR CARDS OF ID (WORD LENGTH 4) 16
C * NCONF CONFIGURATION NUMBER 17
C * RPM SPEED IN RPM 18
C * PCS PERCENT SPEED 19
C * NF NUMBER OF FREQUENCY BANDS 20
C * NM NUMBER OF ANGLES 21
C * NB 1/NB-OCTAVE BANDS 22
C * PWL OVERALL ACOUSTIC POWER LEVEL 23
C * SUMN DECIBEL SUM OF NORMALIZED POWER SPECTRUM 24
C * PSM(27) NORMALIZED POWER SPECTRUM 25
C * DT ANGLE INCREMENT 26
C * AI(19) ANGLES 27
C * NFIL(27) BAND CENTER FREQUENCIES 28
C * DI(27,19) DIRECTIVITY INDEX 29
C * 30
C * * * * * 31
C DIMENSION BUF(27,19),TFA(27,19),SL(27,19) 32
C DIMENSION AM(19) 33
C DIMENSION B(27),C(27),D(27),E(27),SLR(27) 34
C DIMENSION US(6),SD(5) 35
C 36
C DEFINE NUMBER OF SIDELINE DISTANCES 37
C 38
C READ (5,1) RSTD,(SD(I),I=1,5) 39
C FORMAT (6F6.0) 40
C * * * * * 41
C * 42
C * INPUT DATA REQUIRED 43
C * 44
C * ONE CARD WITH UP TO FIVE SIDELINE DISTANCES, 45
C * OR A BLANK CARD FOR NO SIDELINE EXTRAPOLATIONS. 46
C * 47
C * * * * * 48
C DO 2 I=1,5 49
C IF (SD(I).LE.0.0) GO TO 3 50
C 2 51
C CONTINUE 52
C NR=5 52
C GO TO 4 53
C 3 54
C CONTINUE 54
C NR=I-1 55
C CONTINUE 56
C 4 57
C 5 58
C CALL WDATA 58
C * * * * * 59
C * INPUT DATA REQUIRED 60
C * 61
C * ONE OR MORE SETS OF WORKING DATA TO BE LISTED. 62
C * 63
C * * * * * 64
C CALL ASMBL (RSTD,SL) 65
C 66
C 67
C TITLE PAGE 68
C 69
C WRITE (6,6) 70
C 6 FORMAT (1H1) 71
C WRITE (6,7) 72
C 7 FORMAT (1H4,46X,35HN O I S E DATA LISTING////52X,26HC 73
C 10MPUTED FROM WORKING DATA//64X,2HOF//) 74

```



	DO 8 J=1,4	75
8	WRITE (6,9) (A(I,J),I=1,20)	76
9	FORMAT (30X,20A4)	77
	WRITE (6,10)	78
10	FORMAT (/)	79
	WRITE (6,11) NCONF,PCS,RPM,NF,NB,NFIL(1),NFIL(NF),NM,DT,AI(1),AI(N	80
	1M)	81
11	FORMAT (1H ,31X,13HCONFIGURATION,I4,10X,F5.1,14H PERCENT SPEED,10X	82
	1,F8.0,4H RPM//43X,I2,5H - 1/,I1,18H OCTAVE BANDS FROM,I5,3H TO,I6,	83
	26H HERTZ//42X,I3,13H ANGLES EVERY,F4.0,13H DEGREES FROM,F5.0,3H TO	84
	3,F5.0)	85
C		86
C	PAGE THREE	87
C	REFERRED ARRAY	88
C		89
	CALL TITLE (A,NCONF,RPM,PCS)	90
	WRITE (6,12) RSTD	91
12	FORMAT (52X,27H R E F E R R E D A R R A Y//38X,7HDATA AT,F6.1,42H	92
	1 FT RADIUS WITH NO ATMOSPHERIC ATTENUATION/45X,40H(FOR POWER AND D	93
	ZIRECTIVITY COMPUTATIONS)//)	94
	CALL ANGLE (AI,NM)	95
	CALL OASPL (SL,NM,NF,E)	96
	WRITE (6,13) (E(I),I=1,NM)	97
13	FORMAT (15H COMPUTED OASPL,19F6.1)	98
	WRITE (6,10)	99
	CALL TBLOP (SL,NF,NFIL,NM,1)	100
C		101
C		102
C	PAGE FOUR	103
C	ACOUSTIC POWER	104
C		105
C	COMPUTE TOTAL POWER, WATTS	106
C		107
	W=1.0E-13*10.0**((PWL/10.0)	108
	CALL AVSLR (E,AI,DT,NM,SLR0)	109
C		110
C	COMPUTE POWER SPECTRUM	111
C		112
	DO 14 I=1,NF	113
14	C(I)=PWL-SUMN+PSM(I)	114
	DO 16 I=1,NF	115
	DO 15 J=1,NM	116
15	D(J)=SL(I,J)	117
16	CALL AVSLR (D,AI,DT,NM,SLR(I))	118
	CALL TITLE (A,NCONF,RPM,PCS)	119
	WRITE (6,17) RSTD	120
17	FORMAT (40X,53H A C O U S T I C P O W E R C O M P U T A T I O N	121
	1 S//,73X,10HNORMALIZED,10X,6HSIMPLE/29X,4HBAND,8X,9HFREQUENCY,4X,3	122
	21HPower SPECTRUM POWER SPECTRUM,5X,15HSOURCE SPL, R =,F6.1,3H FT	123
	3//)	124
	DO 18 I=1,NF	125
18	WRITE (6,19) I,NFIL(I),C(I),PSM(I),SLR(I)	126
19	FORMAT (30X,I2,I16,3F17.1)	127
	WRITE (6,20) SUMN,SLR0	128
20	FORMAT (/75X,F7.1,8H OVERALL,F9.1,8H OVERALL)	129
	WRITE (6,21)	130
21	FORMAT (/55X,20HTOTAL ACOUSTIC POWER/72X,3H-13)	131
	WRITE (6,22) PWL,W	132
22	FORMAT (52X,5HPWL =,F6.1,16H DB RE 10 WATT//58X,3HW =,F7.1,6H WA	133
	ITTS/)	134
C		135
C		136
C	PAGE FIVE	137

C	NORMALIZED POWER SPECTRUM (GRAPH)	138
C		139
	CALL TITLE (A,NCONF,RPM,PCS)	140
	CALL GRAPH (PSM,NF)	141
C		142
C		143
C	PAGE SIX	144
C	DIRECTIVITY INDEX	145
C		146
	CALL TITLE (A,NCONF,RPM,PCS)	147
	WRITE (6,23)	148
23	FORMAT (50X,33HD I R E C T I V I T Y I N D E X//)	149
	CALL ANGLE (AI,NM)	150
	CALL TBLOP (DI,NF,NFIL,NM,1)	151
	CALL AVSLR (E,AI,DI,NM,DIO)	152
	DO 24 J=1,NM	153
24	E(J)=E(J)-DIO	154
	WRITE (6,25) (E(J),J=1,NM)	155
25	FORMAT (8X,7HVERALL,19F6.1)	156
C		157
C		158
C	PAGE SEVEN	159
C	ATMOSPHERIC ATTENUATION	160
C		161
C	COMPUTE THOUSAND FOOT EXCESS ATTENUATION	162
C		163
	DO 26 J=1,NM	164
26	CALL BASPAT (SL(1,J),RSTD,NF,NB,59.0,70.0,TFA(1,J),RSTD,8)	165
C		166
	CALL TITLE (A,NCONF,RPM,PCS)	167
	WRITE (6,27)	168
27	FORMAT (43X,45H A T M O S P H E R I C A T T E N U A T I O N//35X,	169
	16IHSTANDARD DAY EXCESS ATMOSPHERIC ATTENUATION PER THOUSAND FEET//	170
	237X,56HCOMPUTED FROM REFERRED ARRAY CONSIDERING SPECTRUM SHAPES//)	171
	CALL ANGLE (AI,NM)	172
	CALL TBLOP (TFA,NF,NFIL,NM,1)	173
C		174
C		175
C	PAGE EIGHT	176
C	STANDARD DAY DATA ATMOSPHERIC ABSORPTION	177
C		178
	CALL TITLE (A,NCONF,RPM,PCS)	179
	WRITE (6,28) RSTD	180
28	FORMAT (16X,95HS T A N D A R D D A Y D A T A E X C E S S A	181
	1 T M O S P H E R I C A T T E N U A T I O N//26X,60HADJUSTMENTS T	182
	20 REFERRED ARRAY TO OBTAIN STANDARD DAY DATA AT,F6.0,10H FT RADIUS	183
	3//)	184
	DO 29 J=1,NM	185
	DO 29 I=1,NF	186
29	BUF(I,J)=TFA(I,J)*RSTD/1000.0	187
	CALL ANGLE (AI,NM)	188
	CALL TBLOP (BUF,NF,NFIL,NM,1)	189
C		190
C		191
C	PAGE NINE	192
C	STANDARD DAY DATA	193
C		194
	CALL TITLE (A,NCONF,RPM,PCS)	195
	WRITE (6,30) RSTD	196
30	FORMAT (48X,33HS T A N D A R D D A Y D A T A//43X,7HDATA AT,F6	197
	1.1,30H FT RADIUS ON 59F, 70PC RH DAY//)	198
	DO 31 J=1,NM	199
	DO 31 I=1,NF	200

31	BUF(I,J)=SL(I,J)-BUF(I,J)	201
	CALL ANGLE (AI,NM)	202
	CALL QASPL (BUF,NM,NF,B)	203
	WRITE (6,13) (B(I),I=1,NM)	204
	WRITE (6,10)	205
	CALL TBLOP (BUF,NF,NFIL,NM,1)	206
	WRITE (6,13)	207
	WRITE (6,32) RSTD	208
32	FORMAT (44X,18HPERCEIVED NOISE ON,F8.1,17H FT RADIUS, PNDB//)	209
	CALL ANGLE (AI,NM)	210
	CALL APNDB (BUF,NB,NM,B)	211
	WRITE (6,33) (B(I),I=1,NM)	212
33	FORMAT (15X,19F6.1)	213
C		214
C		215
C	PAGE TEN AND FOLLOWING	216
C	SIDELINE EXTRAPOLATED DATA	217
C		218
C	DELETE ON-AXIS DATA	219
C		220
	MM=NM	221
	DO 34 J=1,MM	222
34	AM(J)=AI(J)	223
	IF (AM(1).GT.0.0) GO TO 36	224
	MM=MM-1	225
	DO 35 J=1,MM	226
	AM(J)=AM(J+1)	227
	DO 35 I=1,NF	228
35	SL(I,J)=SL(I,J+1)	229
36	IF (AM(MM).LT.180.0) GO TO 37	230
	MM=MM-1	231
37	CONTINUE	232
C		233
	KK=NR+1	234
	DO 38 I=2,KK	235
38	DS(I)=SD(I-1)	236
	DS(1)=RSTD	237
	DO 44 K=1,KK	238
	CALL TITLE (A,NCONF,RPM,PCS)	239
	WRITE (6,39) DS(K)	240
39	FORMAT (40X,52HS I D E L I N E E X T R A P O L A T E D D A T A	241
	1 //23X,20HSTANDARD DAY DATA ON,F7.0,59H FT SIDELINE, INCORPORATIN	242
	2G EXCESS ATMOSPHERIC ATTENUATION//)	243
	CALL SIDLAT (MM,AM,RSTD,DS(K),B,C)	244
	DO 40 J=1,MM	245
	DO 40 I=1,NF	246
40	BUF(I,J)=SL(I,J)-B(J)-TFA(I,J)*C(J)/1000.0	247
	CALL ANGLE (AM,MM)	248
	CALL QASPL (BUF,MM,NF,C)	249
	WRITE (6,13) (C(I),I=1,MM)	250
	WRITE (6,10)	251
	CALL TBLOP (BUF,NF,NFIL,MM,0)	252
	WRITE (6,10)	253
	WRITE (6,41) DS(1),DS(K),(B(I),I=1,MM)	254
41	FORMAT (30X,35HINVERSE SQUARE LAW ATTENUATION, FROM,F7.0,14H FT RA	255
	DIUS TO,F7.0,13H FT SIDELINE//15X,19F6.1//)	256
	WRITE (6,10)	257
	CALL APNDB (BUF,NB,NM,B)	258
	WRITE (6,42) DS(K)	259
42	FORMAT (45X,18HPERCEIVED NOISE ON,F7.0,19H FT SIDELINE, PNDB//)	260
	CALL ANGLE (AM,MM)	261
	WRITE (6,43) (B(I),I=1,MM)	262
43	FORMAT (15X,19F6.1)	263

```

IF (NR.LE.0) GO TO 45
44 CONTINUE
45 WRITE (6,6)
C
GO TO 5
END

```

264  
265  
266  
267  
268  
269-

```

SUBROUTINE OASPL (A,NM,NF,OA)
/OASPL - ARRAY OVERALL SOUND PRESSURE LEVELS/
C * * * * *
C *
C * UTILITY ROUTINE TO COMPUTE OVERALL SOUND PRESSURE LEVELS
C * FOR ALL ANGLES OF A DATA ARRAY.
C *
C * A DATA ARRAY
C * NM NUMBER OF ANGLES
C * NF NUMBER OF FREQUENCY BANDS
C * OA OVERALL SPLS FOR ALL ANGLES
C *
C * CALLS DBSUM
C *
C * * * * *
DIMENSION A(27,19),OA(19)
DO 1 J=1,NM
CALL DBSUM (A(1,J),NF,SUM)
1 OA(J)=SUM
RETURN
END

```

1  
2  
3  
4  
5  
6  
7  
8  
9  
10  
11  
12  
13  
14  
15  
16  
17  
18  
19  
20  
21-

```

SUBROUTINE PNDB (NB,LB,PNL,NC)
/PNDB - PERCEIVED NOISE DECIBELS/
C * * * * *
C *
C * COMPUTES PERCEIVED NOISE LEVEL (PNL) IN PNDB IN ACCORDANCE WITH
C * SOCIETY OF AUTOMOTIVE ENGINEERS AEROSPACE RECOMMENDED PRACTICE
C * NO. 365A, AUGUST 15, 1969.
C *
C * NB 1/NB-OCTAVE BANDS
C *
C * LB ARRAY OF 8 OR 24 CONSECUTIVE 1/NB-OCTAVE BAND SOUND
C * PRESSURE LEVELS. LB(1) CONTAINS SPL FOR 63 HZ BAND
C * IF NB = 1. LB(1) CONTAINS SPL FOR 50 HZ BAND IF
C * NB = 3.
C *
C * PNL PERCEIVED NOISE LEVEL IN PNDB
C *
C * NC OUTPUT CONTROL. IF NC EQUALS ZERO, NO PRINTED OUTPUT
C * (OTHER THAN ERROR MESSAGES) WILL BE GENERATED.
C * IF NC IS OTHER THAN ZERO, A TABULATION OF BAND NUM-
C * BERS, LEVELS, NOY VALUES, AND PERCEIVED NOISE LEVEL
C * WILL BE PRINTED.
C *
C * * * * *

```

1  
2  
3  
4  
5  
6  
7  
8  
9  
10  
11  
12  
13  
14  
15  
16  
17  
18  
19  
20  
21  
22  
23  
24

	REAL LB(27),NOY(24),K,NMAX,NBAR,MJ,LK	25
	REAL L(24,5),M(24,4)	26
	DATA L/49.0,44.0,39.0,34.0,30.0,27.0,24.0,21.0,18.0,5*16.0,15.0,	27
	112.0,9.0,5.0,4.0,5.0,6.0,10.0,17.0,21.0,55.0,51.0,46.0,42.0,39.0,	28
	236.0,33.0,30.0,27.0,5*25.0,23.0,21.0,18.0,15.0,2*14.0,15.0,17.0,	29
	323.0,29.0,64.0,60.0,56.0,53.0,51.0,48.0,46.0,44.0,42.0,5*40.0,	30
	438.0,34.0,32.0,30.0,2*29.0,30.0,31.0,37.0,41.0,52.0,51.0,49.0,	31
	547.0,46.0,45.0,43.0,42.0,41.0,5*40.0,38.0,34.0,32.0,30.0,	32
	62*29.0,30.0,31.0,34.0,37.0,91.01,85.88,87.32,79.85,79.76,75.96,	33
	773.96,74.91,94.63,13*100.00,44.29,50.72/	34
	DATA M/0.079520,2*0.068160,0.059640,10*0.053013,0.059640,	35
	12*0.053013,2*0.047712,2*0.053013,0.068160,0.079520,0.0596401,	36
	22*0.058098,0.052288,0.047534,2*0.043573,0.040221,0.037349,	37
	37*0.034859,0.040221,0.037349,4*0.034859,2*0.037349,0.043573,	38
	40.043478,0.040570,2*0.036831,0.035336,2*0.033333,0.032051,	39
	50.030675,6*0.030103,7*0.029960,2*0.042285,15*0.030103,9*0.029960/	40
	IF (NB.EQ.1) GO TO 2	41
	IF (NB.EQ.3) GO TO 3	42
	WRITE (6,1)	43
1	FORMAT (1H0,23HPNDB SUBROUTINE MESSAGE/41H FREQUENCY BANDWIDTH IMP	44
	ROPERLY SPECIFIED/47H RETURN TO CALLING PROGRAM WITH PNDB EQUAL ZE	45
	2RO//)	46
	PNL=0.0	47
	RETURN	48
2	NF=8	49
	K=0.3	50
	MM=3	51
	LL=1	52
	GO TO 4	53
3	NF=24	54
	K=0.15	55
	MM=1	56
	LL=0	57
4	CONTINUE	58
	NMAX=0.0	59
	SUMN=0.0	60
	DO 12 I=1,NF	61
	J=MM*I-LL	62
	IF (LB(I).LT.L(J,1)) GO TO 6	63
	IF (L(J,1).LE.LB(I).AND.LB(I).LT.L(J,2)) GO TO 7	64
	IF (L(J,2).LE.LB(I).AND.LB(I).LT.L(J,3)) GO TO 8	65
	IF (L(J,3).LE.LB(I).AND.LB(I).LT.L(J,5)) GO TO 9	66
	IF (L(J,5).LE.LB(I).AND.LB(I).LT.150.0) GO TO 10	67
	IF (LB(I).EQ.150.0) GO TO 10	68
	WRITE (6,5) LB(I)	69
5	FORMAT (30H PNDB SUBROUTINE ERROR MESSAGE//F6.1,68H DB EXCEEDS RAN	70
	IGE FOR VALID PNDB CALCULATION. RETURN WITH PNDB = 0.)	71
	PNL=0.0	72
	RETURN	73
6	NOY(I)=0.0	74
	GO TO 12	75
7	MJ=M(J,1)	76
	LK=L(J,1)	77
	A=0.1	78
	GO TO 11	79
8	MJ=M(J,2)	80
	LK=L(J,3)	81
	A=1.0	82
	GO TO 11	83
9	MJ=M(J,3)	84
	LK=L(J,3)	85
	A=1.0	86

```

10 GO TO 11
   MJ=M(J,4)
   LK=L(J,4)
   A=1.0
11 NOY(I)=A*10.0**(MJ*(LB(I)-LK))
   IF (NOY(I).GT.NMAX) NMAX=NOY(I)
12 SUMN=SUMN+NOY(I)
   CONTINUE
   NBAR=NMAX+K*(SUMN-NMAX)
   PNL=40.0+33.22*ALOG10(NBAR)
   IF (NC.EQ.0) RETURN
   WRITE (6,13)
13 FORMAT (1H,22HPNDB, SUBROUTINE OUTPUT)
   GO TO (15,22,20),NF
14 WRITE (6,16) (I,I=1,NF)
15 WRITE (6,16) (I,I=1,NF)
16 FORMAT (1H,5H PAND,16,11F7)
   WRITE (6,17) (LB(I),I=1,NF)
17 FORMAT (1H,5HLEVEL,12F7.1)
   WRITE (6,18) PNL
18 FORMAT (1H,92X,F6.1,5H PNDB)
   WRITE (6,19) (NOY(I),I=1,NF)
19 FORMAT (1H,5H NOYS,12F7.1)
   RETURN
20 IF (NF.LE.12) GO TO 14
   WRITE (6,16) (I,I=1,12)
   WRITE (6,17) (LB(I),I=1,12)
   WRITE (6,19) (NOY(I),I=1,12)
   WRITE (6,21)
21 FORMAT (/)
   WRITE (6,16) (I,I=13,NF)
   WRITE (6,17) (LB(I),I=13,NF)
   WRITE (6,18) PNL
   WRITE (6,19) (NOY(I),I=13,NF)
22 RETURN
   END

```

```

SUBROUTINE PNLT (SL,PDB,DBT)
/ PNLT - TONE-CORRECTED PERCEIVED NOISE LEVEL/
* * * * *
* COMPUTES TONE-CORRECTED PERCEIVED NOISE LEVEL (PNLT) FOR A
* 1/3-OCTAVE BAND SOUND SPECTRUM IN ACCORDANCE WITH FEDERAL
* AVIATION REGULATIONS, VOL. III, PART 36 - NOISE STANDARDS,
* AIRCRAFT TYPE CERTIFICATION.
* ALSO COMPUTES PERCEIVED NOISE LEVEL.
*
* SL      1/3-OCTAVE BAND SOUND SPECTRUM
* PDB     PERCEIVED NOISE LEVEL, PNDB
* DBT     TONE-CORRECTED PERCEIVED NOISE LEVEL, PNLT
*
* CALLS   PNDB
*
* * * * *
DIMENSION SL(24),SLP(24),F(24)
DIMENSION SBAR(25)
REAL M(25)
LOGICAL N(24)

```

	INTEGER FREQ(24)	22
	DATA FREQ/50,63,80,100,125,160,200,250,315,400,500,630,800,1000,	23
1	1 1250,1600,2000,2500,3150,4000,5000,6300,8000,10000/	24
	DO 1 I=1,24	25
	IF (SL(I).LT.0.0) SL(I)=0.0	26
1	N(I)=.FALSE.	27
	DO 2 I=4,24	28
2	M(I)=SL(I)-SL(I-1)	29
	DO 3 I=5,24	30
	IF (ABS(M(I)-M(I-1)).LE.5.0) GO TO 3	31
	IF (M(I).GT.0.0.AND.M(I).GT.M(I-1)) N(I)=.TRUE.	32
	IF (M(I).LE.0.0.AND.M(I-1).GT.0.0) N(I-1)=.TRUE.	33
3	CONTINUE	34
	DO 6 I=1,24	35
	IF (.NOT.N(I)) GO TO 4	36
	IF (I.EQ.24) GO TO 5	37
	SLP(I)=(SL(I-1)+SL(I+1))/2.0	38
	GO TO 6	39
4	SLP(I)=SL(I)	40
	GO TO 6	41
5	SLP(24)=SL(23)+M(24)	42
6	CONTINUE	43
	DO 7 I=4,24	44
7	M(I)=SLP(I)-SLP(I-1)	45
	M(3)=M(4)	46
	M(25)=M(24)	47
	DO 8 I=3,23	48
8	SBAR(I)=(M(I)+M(I+1)+M(I+2))/3.0	49
	SLP(3)=SL(3)	50
	DO 9 I=4,24	51
9	SLP(I)=SLP(I-1)+SBAR(I-1)	52
	TMAX=0.0	53
	T=0.0	54
	DO 14 I=3,24	55
	F(I)=SL(I)-SLP(I)	56
	IF (F(I).LE.0.0) GO TO 14	57
	IF (50.LE.FREQ(I).AND.FREQ(I).LT.500) GO TO 10	58
	IF (5000.LE.FREQ(I).AND.FREQ(I).LE.10000) GO TO 10	59
	GO TO 11	60
10	IF (F(I).LT.3.0) T=0.0	61
	IF (3.0.LE.F(I).AND.F(I).LT.20.0) T=F(I)/6.0	62
	IF (20.0.LE.F(I)) T=3.0+1.0/3.0	63
	GO TO 13	64
11	IF (500.LE.FREQ(I).AND.FREQ(I).LE.5000) GO TO 12	65
	GO TO 14	66
12	IF (F(I).LT.3.0) T=0.0	67
	IF (3.0.LE.F(I).AND.F(I).LT.20.0) T=F(I)/3.0	68
	IF (20.0.LE.F(I)) T=6.0+2.0/3.0	69
13	IF (T.GE.TMAX) TMAX=T	70
14	CONTINUE	71
	CALL PNDB (3,SL,PDB,0)	72
	DBT=PDB+TMAX	73
	RETURN	74
	END	75-

```

SUBROUTINE POWER (A,R,AI,NM,DT,T,BAR,PWL,W)
/POWER - TOTAL ACOUSTIC POWER/
* * * * *
C * COMPUTES TOTAL ACOUSTIC POWER BY INCREMENTAL AREA SUMMATION
C * FOR A SET OF ANGLES AND REFERRED SOUND PRESSURE LEVELS ON AN
C * ARC. PRESENCE OF PERFECTLY REFLECTIVE GROUND PLANE ASSUMED.
C *
C * A      ARRAY OF REFERRED SPL DATA ON AN ARC
C * R      ARC RADIUS
C * AI     CONSECUTIVE ANGLES CORRESPONDING WITH ELEMENTS OF A
C * NM     NUMBER OF ANGLES
C * DT     ANGLE INCREMENT
C * T      TEMPERATURE (DEGREES FAHRENHEIT)
C * BAR    BAROMETER (INCHES HG)
C * PWL    POWER LEVEL RE 0.1 PICOWATT
C * W      ACOUSTIC POWER, WATTS
* * * * *
DIMENSION A(27),AI(19)
CONST=2.0*59.141053*1.0E-15
SPHERE=4.0*3.1415927*R**2
C=49.02*SQRT(T+459.67)
RHO=0.0023769*518.688/(T+459.67)*BAR/29.92
F=3.1415927/180.0
NFT=1
NLT=NM
SUM=0.0
IF (AI(1).GT.0.0) GO TO 1
SUM=SUM+10.0**(A(1)/10.0)
NFT=2
1 IF (AI(NM).LT.180.0) GO TO 2
SUM=SUM+10.0**(A(NM)/10.0)
NLT=NM-1
2 SUM=TAN(DT/4.0*F)*SUM/2.0
DO 3 J=NFT,NLT
3 SUM=SUM+10.0**(A(J)/10.0)*SIN(AI(J)*F)
W=CONST/(RHO*C)*SPHERE*SIN(DT/2.0*F)*SUM
PWL=130.0+10.0*ALOG10(W)
RETURN
END

```

```

SUBROUTINE SIDLAT (NM,AI,RSTD,SIDIST,SINAT,RADIST)
/SIDLAT - SIDELINE ATTENUATION/
* * * * *
C * UTILITY ROUTINE TO COMPUTE SIDELINE RADIAL DISTANCES AND
C * ATTENUATIONS FOR ALL ANGLES.
C *
C * NM      NUMBER OF ANGLES
C * AI      ANGLES
C * RSTD    ARC RADIUS ABOUT SOURCE
C * SIDIST  NORMAL DISTANCE, SOURCE CENTERLINE TO PARALLEL
C *         SIDELINE
C * SINAT   INVERSE SQUARE ATTENUATIONS IN DECIBELS,
C *         ARC TO SIDELINE
C * RADIST  RADIAL DISTANCES, SOURCE TO SIDELINE
* * * * *

```



```

DIMENSION AI(19),SINAT(19),RADIST(19)
F=3.1415927/180.0
RSQ=20.0*ALOG10(SIDIST/RSTD)
DO 2 J=1,NM
ST=SIN(AI(J)*F)
IF (ST.LE.0.0) GO TO 1
SINAT(J)=RSQ-20.0*ALOG10(ST)
RADIST(J)=SIDIST/ST
GO TO 2
1 SINAT(J)=0.0
  RADIST(J)=0.0
2 CONTINUE
  RETURN
  END

```

```

SUBROUTINE TABLE (RSTD,SD)
/TALE - TABLE OF DATA FOR REPORTING/
* * * * *
C * READS WORKING DATA AND PREPARES A TABLE OF NOISE DATA IN A
C * FORMAT SUITABLE FOR REPORTING PURPOSES. DATA INCLUDES
C * STANDARD-DAY 1/3-OCTAVE BAND AND OVERALL SOUND PRESSURE
C * LEVELS ON AN ARC, OVERALL POWER LEVEL AND POWER SPECTRUM,
C * CORRESPONDING SIMPLE SOURCE SOUND PRESSURE LEVELS, AND
C * OPTIONAL PERCEIVED NOISE LEVELS ON SELECTED SIDELINES.
C *
C * NOTE
C * PROGRAM LOGIC AND FORMAT STATEMENTS CODED FOR NB = 3, AND
C * NM = 16. CODING MUST BE REVISED FOR OTHER CONDITIONS.
C *
C * RSTD RADIUS FOR WHICH DATA TO BE PREPARED
C * SD OPTIONAL SIDELINE DISTANCES FOR PERCEIVED NOISE
C *
C * CALLS ASMBL, AVSLR, DBSUM, FARDTA, PNDB, WDATA
C *
C * * * * *
COMMON /WD/A(20,4),NCONF,RPM,PCS,NF,NM,NB,PWL,SUMN,PSM(27),DT,
1 AI(19),NFIL(27),DI(27,19)
C * * * * *
C * VARIABLES IN COMMON BLOCK /WD/
C *
C * A(20,4) FOUR CARDS OF ID (WORD LENGTH 4)
C * NCONF CONFIGURATION NUMBER
C * RPM SPEED IN RPM
C * PCS PERCENT SPEED
C * NF NUMBER OF FREQUENCY BANDS
C * NM NUMBER OF ANGLES
C * NB 1/NB-OCTAVE BANDS
C * PWL OVERALL ACOUSTIC POWER LEVEL
C * SUMN DECIBEL SUM OF NORMALIZED POWER SPECTRUM
C * PSM(27) NORMALIZED POWER SPECTRUM
C * DT ANGLE INCREMENT
C * AI(19) ANGLES
C * NFIL(27) BAND CENTER FREQUENCIES
C * DI(27,19) DIRECTIVITY INDEX
C *
C * * * * *

```

```

DIMENSION SL(27,19),OASPL(19),SLR(27,19),PW(27),AVSPL(27),
1SLS(27,19),PNL(19),B(19),OASL(19)
DIMENSION IAI(19),SD(5)
C
CALL WDATA
C * * * * *
C *
C * INPUT DATA REQUIRED
C *
C * ONE SET OF WORKING DATA FOR EACH SUBROUTINE CALL.
C *
C * * * * *
CALL ASMBL (RSTD,SL)
DO 2 I=1,NF
DO 1 J=1,NM
1 B(J)=SL(I,J)
2 CALL AVSLR (B,AI,DT,NM,AVSPL(I))
DO 3 J=1,NM
3 CALL DBSUM (SL(I,J),NF,OASL(J))
CALL AVSLR (OASL,AI,DT,NM,AVGOA)
CALL FARDTA (SL,RSTD,NF,NB,NM,AI,59.0,70.0,RSTD,0,SLR)
DO 4 J=1,NM
4 CALL DRSUM (SLR(I,J),NF,OASPL(J))
DO 5 I=1,NF
5 PW(I)=PWL-SUMN+PSM(I)
WRITE (6,6)
6 FORMAT (1H0,23X,87HDATA ADJUSTED TO STANDARD DAY OF 15 DEGREES C,
170 PERCENT RELATIVE HUMIDITY
WRITE (6,7)
7 FORMAT (1H ,35X,20HSPL RE .G0G02 N/SQ M,10X,18HPWL RE .1 PICOWATT)
DO 8 I=1,NM
8 IAI(I)=AI(I)
WRITE (6,9)
9 FORMAT (1H0,1X,9HFREQUENCY,46X,10HANGLE, DEG,45X,6HSIMPLE,2X,5HPOW
1ER/112X,6HSOURCE,2X,5HLEVEL)
WRITE (6,10) (IAI(I),I=1,NM)
10 FORMAT (1H ,10X,I7I6)
RMETER=RSTD*0.3048
WRITE (6,11) RMETER
11 FORMAT (1H+,113X,3HSPL,3X,5H(PWL)//30X,46H1/3-OCTAVE BAND SOUND PR
1ESSURE LEVELS (SPL) ON,F6.1,13H METER RADIUS//)
DO 13 I=1,NF
WRITE (6,12) NFIL(I),(SLR(I,J),J=1,NM),AVSPL(I),PW(I)
12 FORMAT (1H ,1X,I8,2X,16F6.1,1X,2F8.1)
13 IF (MOD(I,3).EQ.0) WRITE (6,14)
14 FORMAT (1H )
WRITE (6,15) (OASPL(J),J=1,NM),AVGOA,PWL
15 FORMAT (1H ,3X,7HOVERALL,1X,16F6.1,1X,2F8.1)
DO 16 I=1,5
IF (SD(I).LE.0.0) GO TO 17
16 CONTINUE
IS=5
GO TO 18
17 IS=I-1
IF (IS.LE.0) RETURN
18 CONTINUE
WRITE (6,19)
19 FORMAT (1H0,2X,8HDISTANCE,35X,31HSIDELINE PERCEIVED NOISE LEVELS//
1)
DO 21 I=1,IS
CALL FARDTA (SL,RSTD,NF,NB,NM,AI,59.0,70.0,SD(I),1,SLS)
DO 2 J=1,NM

```

```

20 CALL PND8 (NB,SLS(1,J),PNL(J),0) 106
SDM=SU(1)*0.3048 107
21 WRITE (6,22) SDM,(PNL(J),J=1,NM) 108
22 FORMAT (1H ,F6.1,2H M,3X,19F6.1) 109
RETURN 110
END 111-

```

```

SUBROUTINE TBLOP (A,NF,NFIL,NM,NEG) 1
C /TBLOP - TABLE OUTPUT/ 2
C * * * * * 3
C * 4
C * UTILITY ROUTINE TO OUTPUT DATA ARRAY. 5
C * 6
C * A DATA ARRAY 7
C * NF NUMBER OF FREQUENCY BANDS 8
C * NFIL BAND CENTER FREQUENCIES 9
C * NM NUMBER OF ANGLES 10
C * NEG OUTPUT CONTROL. IF NEG EQUALS ZERO, ALL NEGATIVE 11
C * VALUES IN ARRAY A WILL BE REPLACED WITH ZERO FOR 12
C * EASE OF READING. IF NEG IS OTHER THAN ZERO, ARRAY 13
C * IS UNAFFECTED. 14
C * 15
C * * * * * 16
C DIMENSION A(27,19) 17
DIMENSION NFIL(27) 18
IF (NEG.NE.0) GO TO 2 19
DO 1 J=1,NM 20
DO 1 I=1,NF 21
IF (A(I,J).LT.0.0) A(I,J)=0.0 22
1 CONTINUE 23
2 WRITE (6,3) 24
3 FORMAT (15H BAND FREQUENCY) 25
DO 4 I=1,NF 26
4 WRITE (6,5) I,NFIL(I),(A(I,J),J=1,NM) 27
5 FORMAT (1H ,I3,I8,3X,19F6.1) 28
WRITE (6,6) 29
6 FORMAT (//) 30
RETURN 31
END 32-

```

```

SUBROUTINE TITLE (A,NCONF,RPM,PCS) 1
C /TITLE - TITLE OUTPUT/ 2
C * * * * * 3
C * 4
C * UTILITY ROUTINE TO OUTPUT ID INFORMATION. 5
C * 6
C * A FOUR ID CARDS 7
C * NCONF CONFIGURATION NUMBER 8
C * RPM SPEED IN RPM 9
C * PCS PERCENT SPEED 10
C * 11
C * * * * * 12

```

	DIMENSION A(20,4)	13
	WRITE (6,1)	14
1	FORMAT (1H1)	15
2	FORMAT (1H ,20A4)	16
	WRITE (6,3) (A(J,1),J=1,20),NCONF	17
3	FORMAT (1H ,20A4,17X,16HCONFIGURATION NO,15)	18
	WRITE (6,4) (A(J,2),J=1,20),RPM	19
4	FORMAT (1H ,20A4,17X,7HSPEED =,F6.0,4H RPM)	20
	WRITE (6,5) (A(J,3),J=1,20),PCS	21
5	FORMAT (1H ,20A4,17X,15HPERCENT SPEED =,F6.1)	22
	WRITE (6,2) (A(J,4),J=1,20)	23
	WRITE (6,6)	24
6	FORMAT (//)	25
	RETURN	26
	END	27-

	SUBROUTINE TITLE2	1
	/TITLE2 - TITLE OUTPUT/	2
C	* * * * *	3
C	*	4
C	* UTILITY ROUTINE TO OUTPUT ID INFORMATION.	5
C	*	6
C	* * * * *	7
	COMMON A(20,4,2),AI(19,2),NFIL(27,2),NCONF(2),RPM(2),PCS(2),	8
	1 NR(2),NF(2),NM(2),DT(2)	9
	DIMENSION SET(2),AIL(2),IFIL(2)	10
	DATA SET(1),SET(2)/3HONE,3HTWO/	11
	DO 1 K=1,2	12
	J=NF(K)	13
	IFIL(K)=NFIL(J,K)	14
	J=NM(K)	15
1	AIL(K)=AI(J,K)	16
	WRITE (6,2)	17
2	FORMAT (1H ,36X,53HCOMPARISON OF TWO DATA S	18
	LET S////)	19
	DO 3 K=1,2	20
3	WRITE (6,4) SET(K),(A(I,1,K),I=1,20),(A(I,2,K),I=1,20),NCONF(K),PC	21
	1S(K),RPM(K),(A(I,3,K),I=1,20),NF(K),NB(K),NFIL(1,K),IFIL(K),(A(I,4	22
	2,K),I=1,20),NM(K),DT(K),AI(1,K),AIL(K)	23
4	FORMAT (10H DATA SET ,A3,/1H ,20A4,5X,38HCONFIGURATION PERCENT	24
	1 SPEED RPM/1H ,20A4,I12,F20.1,F13.0/1H ,20A4,I7,5H - 1/,I1,18H	25
	2OCTAVE BANDS FROM,I5,3H TO,I6,6H HERTZ/1H ,20A4,I7,13H ANGLES EVER	26
	3Y,F4.0,13H DEGREES FROM,F5.0,3H TO,F5.0//)	27
	RETURN	28
	END	29-

	SUBROUTINE WDATA	1
	/WDATA - WORKING DATA/	2
C	* * * * *	3
C	*	4
C	* READS ONE SET OF WORKING DATA FROM CARDS INTO STORAGE	5
C	* COMMON BLOCK /WD/.	6
C	*	7
C	* * * * *	8

```

COMMON /WD/A(20,4),NCONF,RPM,PCS,NF,NM,NB,PWL,SUMN,PSM(27),DT,
1 AI(19),NFIL(27),DI(27,19)
C * * * * * 9
C * 10
C * 11
C * 12
C * VARIABLES RESULTING IN COMMON BLOCK /WD/ 13
C * 14
C * A(20,4) FOUR CARDS OF ID (WORD LENGTH 4) 15
C * NCONF CONFIGURATION NUMBER 16
C * RPM SPEED IN RPM 17
C * PCS PERCENT SPEED 18
C * NF NUMBER OF FREQUENCY BANDS 19
C * NM NUMBER OF ANGLES 20
C * NB 1/NB-OCTAVE BANDS 21
C * PWL OVERALL ACOUSTIC POWER LEVEL 22
C * SUMN DECIBEL SUM OF NORMALIZED POWER SPECTRUM 23
C * PSM(27) NORMALIZED POWER SPECTRUM 24
C * DT ANGLE INCREMENT 25
C * AI(19) ANGLES 26
C * NFIL(27) BAND CENTER FREQUENCIES 27
C * DI(27,19) DIRECTIVITY INDEX 28
C * 29
C * * * * * 30
DO 1 J=1,4 31
1 READ (5,2) (A(I,J),I=1,20) 32
2 FORMAT (20A4) 33
3 READ (5,3) NCONF,RPM,PCS,NF,NM,NB 34
4 FORMAT (I4,2F8.1,3I3) 35
IF (NF.EQ.10) GO TO 5 36
5 READ (5,4) PWL,SUMN,(PSM(I),I=1,NF) 37
6 FORMAT (12F6.1/(12X,10F6.1)) 38
GO TO 7 39
7 READ (5,6) PWL,SUM,(PSM(I),I=1,NF) 40
8 FORMAT (12F6.1) 41
9 READ (5,6) DT,(AI(J),J=1,NM) 42
DO 11 I=1,NF 43
IF (NM.EQ.10) GO TO 9 44
8 READ (5,8) NFIL(I),(DI(I,J),J=1,NM) 45
9 FORMAT (6X,I6,10F6.1/(12X,10F6.1)) 46
GO TO 11 47
9 READ (5,10) NFIL(I),(DI(I,J),J=1,NM) 48
10 FORMAT (6X,I6,10F6.1) 49
11 CONTINUE 50
RETURN 51
END 52-

```

```

SUBROUTINE WODAG 1
C /WODAG - WORKING DATA GENERATION/ 2
C * * * * * 3
C * 4
C * STANDARDIZES MEASURED DATA. PREPARES DATA LISTINGS. PUNCHES 5
C * WORKING DATA. 6
C * 7
C * CALLS ANGLE, APNDB, AVSLR, HASPAT, DBSUM, GRAPH, 8
C * OASPL, POWER, SIDLAT, TBLOP, TITLE 9
C * 10
C * * * * * 11
COMMON /WODA/A(20,4),NCONF,RPM,PCS,SL(27,19),T,RH,BAR,AO,DT,R(19),
1NM,NF,NB,RSTD,SD(5) 12
13

```

```

C * * * * * 14
C * 15
C * VARIABLES NECESSARY IN COMMON BLOCK /WODA/ 16
C * 17
C * A(20,4) FOUR CARDS OF ID (WORD LENGTH 4) 18
C * NCCNF CONFIGURATION NUMBER 19
C * RPM SPEED IN RPM 20
C * PCS PERCENT SPEED 21
C * SL(27,19) MEASURED ARRAY (TEST DAY SOUND PRESSURE LEVELS) 22
C * T TEST TEMPERATURE (DEGREES FAHRENHEIT) 23
C * RH TEST RELATIVE HUMIDITY 24
C * BAR TEST BAROMETER (INCHES HG) 25
C * AO FIRST MICROPHONE ANGLE 26
C * DT MICROPHONE ANGLE INCREMENT 27
C * R(19) MICROPHONE RADII 28
C * NM NUMBER OF MICROPHONES 29
C * NF NUMBER OF FREQUENCY BANDS 30
C * NB KIND OF FRACTIONAL-OCTAVE BANDS (1 OR 3) 31
C * RSTD STANDARD RADIUS FOR DATA LISTINGS 32
C * SD(5) OPTIONAL SIDELINE DISTANCES FOR DATA EXTRAPOLATIONS 33
C * 34
C * * * * * 35
C DIMENSION BUF(27,19),TFA(27,19) 36
C DIMENSION AI(19),AM(19) 37
C DIMENSION B(27),C(27),D(27),E(27),NFIL(27),SLR(27) 38
C DIMENSION DS(6) 39
C INTEGER FREQ(27) 40
C DATA FREQ/50,63,80,100,125,160,200,250,315,400,500,630,800,1000, 41
C 1 1250,1600,2000,2500,3150,4000,5000,6300,8000,10000,12500,16000, 42
C 2 20000/ 43
C 44
C DEFINE NUMBER OF SIDELINE DISTANCES 45
C 46
C DO 1 I=1,5 47
C IF (SD(I).LE.0.0) GO TO 2 48
C CONTINUE 49
C NR=5 50
C GO TO 3 51
C 2 CONTINUE 52
C NR=I-1 53
C 3 CONTINUE 54
C 55
C DEFINE BAND CENTER FREQUENCIES 56
C 57
C DO 4 I=1,NF 58
C J=I 59
C IF (NB.EQ.1) J=3*I-1 60
C NFIL(I)=FREQ(J) 61
C 62
C DEFINE MICROPHONE ANGLES 63
C 64
C DO 5 I=1,NM 65
C AI(I)=AO+DT*FLOAT(I-1) 66
C 67
C SIMPLY ADJUST TO CONSTANT RADIUS FOR REVIEW ONLY 68
C 69
C DO 6 J=1,NM 70
C DL=20.0*ALOG10(R(J)/RSTD) 71
C DO 6 I=1,NF 72
C SL(I,J)=SL(I,J)+DL 73
C 74
C 75
C PAGE ONE 76

```

```

C           MEASURED ARRAY                                77
C                                                    78
WRITE (6,7)                                           79
C                                                    80
7  FORMAT (1H1)                                       81
DO 9 J=1,4                                           81
8  WRITE (6,9) (A(I,J),I=1,20)                       82
9  FORMAT (26X,20A4//)                                83
WRITE (6,10) NCONF,RPM,PCS                           84
10 -FORMAT (16X,16HCONFIGURATION NO,I5,10X,7HSPEED =,F6.0,4H RPM,10X,1
15HPERCENT SPEED =,F6.1//)                            85
WRITE (6,11) T,RH,RAR                                86
11  FORMAT (10X,17HTEST CONDITIONS -,7X,13HTEMPERATURE =,F5.1,2H F,6X,
119HRELATIVE HUMIDITY =,F5.1,3H PC,6X,11HBAROMETER =,F6.2,6H IN HG/
2//)                                                  87
WRITE (6,12) RSTD                                     87
12  FORMAT (52X,27HM E A S U R E D   A R R A Y,//37X,41HMEASURED DATA
1SIMPLY ADJUSTED TO CONSTANT,F6.1,10H FT RADIUS//)    88
CALL ANGLE (AI,NM)                                   88
CALL OASPL (SL,NM,NF,B)                              89
WRITE (6,13) (B(I),I=1,NM)                           90
13  FORMAT (15H COMPUTED OASPL,19F6.1)                91
WRITE (6,14)                                          91
14  FORMAT (/)                                        92
CALL TBLOP (SL,NF,NFIL,NM,1)                          93
WRITE (6,15) (R(I),I=1,NM)                            94
15  FORMAT (48X,31HORIGINAL MICROPHONE RADII, FEET//15X,19F6.1) 95
C                                                    95
C                                                    96
C           PAGE TWO                                    96
C           TEST DAY ATMOSPHERIC ABSORPTION            97
C                                                    97
CALL TITLE (A,NCONF,RPM,PCS)                          98
DO 16 J=1,NM                                          98
CALL BASPAT (SL(1,J),RSTD,NF,NB,T,RH,B,RSTD,C)       99
DO 16 I=1,NF                                          100
BUF(I,J)=B(I)*R(J)/1000.0                            101
16  SL(I,J)=SL(I,J)+BUF(I,J)                          102
WRITE (6,17) T,RH                                    103
17  FORMAT (27X,77H T E S T   D A Y   E X C E S S   A T M O S P H E R I
1 C   A T T E N U A T I O N //34X,62H ADDITIONS TO UNADJUSTED MEASURE
2D DATA TO OBTAIN REFERRED ARRAY//49X,33H TEMPERATURE   RELATIVE H
3UMIDITY/46X,F10.0,2H F,12X,F5.1,3H PC//)            104
CALL ANGLE (AI,NM)                                   105
CALL TBLOP (BUF,NF,NFIL,NM,1)                         106
C                                                    107
C                                                    108
C           PAGE THREE                                  108
C           REFERRED ARRAY                              109
C                                                    109
CALL TITLE (A,NCONF,RPM,PCS)                          110
WRITE (6,18) RSTD                                     110
18  FORMAT (52X,27H R E F E R R E D   A R R A Y//38X,7H DATA AT,F6.1,42H
1 FT RADIUS WITH NO ATMOSPHERIC ATTENUATION/45X,40H (FOR POWER AND D
2IRECTIVITY COMPUTATIONS)//)                          111
CALL ANGLE (AI,NM)                                   112
CALL OASPL (SL,NM,NF,E)                              113
WRITE (6,13) (E(I),I=1,NM)                           114
WRITE (6,14)                                          115
CALL TBLOP (SL,NF,NFIL,NM,1)                          116
C                                                    117
C                                                    118
C           PAGE FOUR                                  118
C           ACOUSTIC POWER COMPUTATIONS                119
C                                                    119

```

C		140
C	COMPUTE TOTAL POWER	141
C		142
	CALL POWER (E,RSTD,AI,NM,DT,T,BAR,PWL,W)	143
	CALL AVSLR (E,AI,DT,NM,SLRO)	144
C		145
C	COMPUTE POWER SPECTRUM	146
C		147
	DO 20 I=1,NF	148
	DO 19 J=1,NM	149
19	B(J)=SL(I,J)	150
	CALL POWER (B,RSTD,AI,NM,DT,T,BAR,PW,D)	151
20	C(I)=PW	152
	CMAX=0.0	153
	DO 21 I=1,NF	154
	IF (C(I).GE.CMAX) CMAX=C(I)	155
21	CONTINUE	156
	DO 22 I=1,NF	157
22	B(I)=C(I)-CMAX	158
	CALL DBSUM (B,NF,SUM)	159
	DO 24 I=1,NF	160
	DO 23 J=1,NM	161
23	D(J)=SL(I,J)	162
24	CALL AVSLR (D,AI,DT,NM,SLR(I))	163
	CALL TITLE (A,NCONF,RPM,PCS)	164
	WRITE (6,25) RSTD	165
25	FORMAT (40X,53HACUSTIC POWER COMPUTATION	166
	1 S//,73X,10HNORMALIZED,10X,6HSIMPLE/29X,4HBAND,8X,9HFREQUENCY,4X,3	167
	21HPOWER SPECTRUM POWER SPECTRUM,5X,15HSOURCE SPL, R =,F6.1,3H FT	168
	3//)	169
	DO 26 I=1,NF	170
26	WRITE (6,27) I,NFIL(I),C(I),B(I),SLR(I)	171
27	FORMAT (30X,12,I16,3F17.1)	172
	WRITE (6,28) SUM,SLRO	173
28	FORMAT (/75X,F7.1,8H OVERALL,F9.1,8H OVERALL)	174
	WRITE (6,29)	175
29	FORMAT (/55X,20HTOTAL ACOUSTIC POWER/72X,3H-13)	176
	WRITE (6,30) PWL,W	177
30	FORMAT (52X,5HPWL =,F6.1,16H DB RE 10 WATT//58X,3HW =,F7.1,6H WA	178
	1TT\$/)	179
C		180
C	PUNCH HEADER AND POWER DATA	181
C		182
	DO 31 J=1,4	183
31	PUNCH 32, (A(I,J),I=1,20)	184
32	FORMAT (20A4)	185
	PUNCH 33, NCONF,RPM,PCS,NF,NM,NB	186
33	FORMAT (I4,2F8.1,3I3)	187
	IF (NF.EQ.10) GO TO 35	188
	PUNCH 34, PWL,SUM,(B(I),I=1,NF)	189
34	FORMAT (12F6.1/(12X,10F6.1))	190
	GO TO 36	191
35	PUNCH 41, PWL,SUM,(B(I),I=1,NF)	192
36	CONTINUE	193
C		194
C		195
C	PAGE FIVE	196
C	NORMALIZED POWER SPECTRUM (GRAPH)	197
C		198
	CALL TITLE (A,NCONF,RPM,PCS)	199
	CALL GRAPH (B,NF)	200



C		201
C		202
C	PAGE SIX	203
C	DIRECTIVITY INDEX	204
C		205
	CALL TITLE (A,NCONF,RPM,PCS)	206
	WRITE (6,37)	207
37	FORMAT (5X,33HD I R E C T I V I T Y I N D E X//)	208
	CALL ANGLE (AI,NM)	209
	DO 38 I=1,NF	210
	DO 38 J=1,NM	211
38	BUF(I,J)=SL(I,J)-SLR(I)	212
	CALL TBLOP (BUF,NF,NFIL,NM,1)	213
	CALL AVSLR (E,AI,DT,NM,DIO)	214
	DO 39 J=1,NM	215
39	E(J)=E(J)-DIO	216
	WRITE (6,40) (E(J),J=1,NM)	217
40	FORMAT (8X,7HOVERALL,19F6.1)	218
C		219
C	PUNCH ANGLES, DIRECTIVITY INDEX	220
C		221
	PUNCH 41, DT,(AI(J),J=1,NM)	222
41	FORMAT (12F6.1)	223
	DO 45 I=1,NF	224
	IF (NM.EQ.10) GO TO 43	225
	PUNCH 42, I,NFIL(I),(BUF(I,J),J=1,NM)	226
42	FORMAT (2I6,10F6.1/12X,10F6.1)	227
	GO TO 45	228
43	PUNCH 44, I,NFIL(I),(BUF(I,J),J=1,NM)	229
44	FORMAT (2I6,10F6.1)	230
45	CONTINUE	231
C		232
C		233
C	PAGE SEVEN	234
C	ATMOSPHERIC ATTENUATION	235
C		236
C	COMPUTE THOUSAND FOOT EXCESS ATTENUATION	237
C		238
	DO 46 J=1,NM	239
46	CALL BASPAT (SL(1,J),RSTD,NF,NB,59.0,70.0,TFA(1,J),RSTD,B)	240
C		241
	CALL TITLE (A,NCONF,RPM,PCS)	242
	WRITE (6,47)	243
47	FORMAT (43X,45H A T M O S P H E R I C A T T E N U A T I O N//35X, 16HSTANDARD DAY EXCESS ATMOSPHERIC ATTENUATION PER THOUSAND FEET// 237X,56HCOMPUTED FROM REFERRED ARRAY CONSIDERING SPECTRUM SHAPES//)	244
	CALL ANGLE (AI,NM)	245
	CALL TBLOP (TFA,NF,NFIL,NM,1)	246
		247
		248
		249
C		250
C		251
C	PAGE EIGHT	252
C	STANDARD DAY DATA ATMOSPHERIC ABSORPTION	253
C		254
	CALL TITLE (A,NCONF,RPM,PCS)	255
	WRITE (6,48) RSTD	256
48	FORMAT (16X,95H S T A N D A R D D A Y D A T A E X C E S S A 1 T M O S P H E R I C A T T E N U A T I O N//26X,60HADJUSTMENTS T 20 REFERRED ARRAY TO OBTAIN STANDARD DAY DATA AT,F6.0,10H FT RADIUS 3//)	257
		258
		259
	DO 49 J=1,NM	260
	DO 49 I=1,NF	261
49	BUF(I,J)=TFA(I,J)*RSTD/1000.0	262
	CALL ANGLE (AI,NM)	263

	CALL TBLOP (BUF,NF,NFIL,NM,1)	264
C		265
C		266
C	PAGE NINE	267
C	STANDARD DAY DATA	268
C		269
	CALL TITLE (A,NCONF,RPM,PCS)	270
	WRITE (6,50) RSTD	271
50	FORMAT (48X,33HS T A N D A R D D A Y D A T A //43X,7HDATA AT,F6	272
	1.1,30H FT RADIUS ON 59F, 70PC RH DAY//)	273
	DO 51 J=1,NM	274
	DO 51 I=1,NF	275
51	BUF(I,J)=SL(I,J)-BUF(I,J)	276
	CALL ANGLE (AI,NM)	277
	CALL DASPL (BUF,NM,NF,B)	278
	WRITE (6,13) (B(I),I=1,NM)	279
	WRITE (6,14)	280
	CALL TBLOP (BUF,NF,NFIL,NM,1)	281
	WRITE (6,14)	282
52	WRITE (6,52) RSTD	283
	FORMAT (44X,18HPERCEIVED NOISE ON,F8.1,17H FT RADIUS, PNDB//)	284
	CALL ANGLE (AI,NM)	285
	CALL APNDB (BUF,NB,NM,B)	286
	WRITE (6,53) (B(I),I=1,NM)	287
53	FORMAT (15X,19F6.1)	288
C		289
C		290
C	PAGE TEN AND FOLLOWING	291
C	SIDELINE EXTRAPOLATED DATA	292
C		293
C	DELETE ON-AXIS DATA	294
C		295
	MM=NM	296
	DO 54 J=1,MM	297
54	AM(J)=AI(J)	298
	IF (AM(1).GT.0.0) GO TO 56	299
	MM=MM-1	300
	DO 55 J=1,MM	301
	AM(J)=AM(J+1)	302
	DO 55 I=1,NF	303
55	SL(I,J)=SL(I,J+1)	304
56	IF (AM(MM).LT.180.0) GO TO 57	305
	MM=MM-1	306
57	CONTINUE	307
C		308
	KK=NR+1	309
	DO 58 I=2,KK	310
58	DS(I)=SD(I-1)	311
	DS(I)=RSTD	312
	DO 64 K=1,KK	313
	CALL TITLE (A,NCONF,RPM,PCS)	314
	WRITE (6,59) DS(K)	315
59	FORMAT (40X,52HS I D E L I N E E X T R A P O L A T E D D A T A	316
	1 //23X,20HSTANDARD DAY DATA ON,F7.0,59H FT SIDELINE, INCORPORATIN	317
	2G EXCESS ATMOSPHERIC ATTENUATION//)	318
	CALL SIDLAT (MM,AM,RSTD,DS(K),B,C)	319
	DO 60 J=1,MM	320
	DO 60 I=1,NF	321
60	BUF(I,J)=SL(I,J)-B(J)-TFA(I,J)*C(J)/1000.0	322
	CALL ANGLE (AM,MM)	323
	CALL DASPL (BUF,MM,NF,C)	324
	WRITE (6,13) (C(I),I=1,MM)	325
	WRITE (6,14)	326

```

CALL TBLOP (BUF,NE,NFIL,MM,0) 327
WRITE (6,14) 328
WRITE (6,61) DS(1),DS(K), (B(I),I=1,MM) 329
61 FORMAT (30X,35HINVERSE SQUARE LAW ATTENUATION FROM,F7.0,14H FT RA 330
1DIUS TO,F7.0,13H FT SIDELINE//15X,19F6.1//) 331
WRITE (6,14) 332
CALL APNOB (BUF,NB,NM,B) 333
WRITE (6,62) DS(K) 334
62 FORMAT (45X,18HPERCEIVED NOISE ON,F7.0,19H FT SIDELINE, PNOB//) 335
CALL ANGLE (AM,MM) 336
WRITE (6,63) (B(I),I=1,MM) 337
63 FORMAT (15X,19F6.1) 338
IF (NR.LE.0) GO TO 65 339
64 CONTINUE 340
65 WRITE (6,7) 341
C 342
RETURN 343
END 344-

```

## REFERENCES

1. Montegani, Francis J.: Noise Generated by Quiet Engine Fans. I - Fan B. NASA TM X-2528, 1972.
2. Anon.: Standard Values of Atmospheric Absorption as a Function of Temperature and Humidity for Use in Evaluating Aircraft Flyover Noise. Aerospace Recommended Practice No. 866, SAE, Aug. 1964.
3. Anon.: Definitions and Procedures for Computing the Perceived Noise Level of Aircraft Noise. Aerospace Recommended Practice No. 865A, SAE, Aug. 1969.
4. Anon.: Federal Aviation Regulations. Vol. III, pt. 36 - Noise Standards: Aircraft Type Certification.

TABLE I. - SAMPLE OUTPUT FROM WORKING DATA GENERATION SUBROUTINE WODAG

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

LEWIS RESEARCH CENTER

PROPULSION SYSTEMS ACOUSTICS BRANCH

SAMPLE NOISE DATA

CONFIGURATION NO 100 SPEED = 1800. RPM PERCENT SPEED = 75.0

TEST CONDITIONS - TEMPERATURE = 50.0 F RELATIVE HUMIDITY = 60.0 PC BAROMETER = 30.00 IN HG

M E A S U R E D A R R A Y

MEASURED DATA SIMPLY ADJUSTED TO CONSTANT 100.0 FT RADIUS

PAND FREQUENCY	MEASURED DATA SIMPLY ADJUSTED TO CONSTANT 100.0 FT RADIUS																
	ANGLE	10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.
COMPLETED CASPL	90.9	92.1	92.6	93.8	91.6	89.8	89.2	89.0	91.8	92.6	93.7	94.8	95.3	94.0	93.6	91.1	
1	50	66.1	64.9	65.9	68.6	67.6	67.2	68.9	66.4	68.7	69.6	70.6	73.1	74.4	77.9	77.9	
2	63	64.3	65.6	65.1	63.8	65.0	65.8	66.1	66.3	66.8	69.3	71.4	74.1	76.1	79.0	80.0	
3	80	65.3	65.8	64.0	64.1	64.8	65.3	65.1	66.8	68.5	70.3	72.8	74.0	77.0	79.3	83.0	83.3
4	100	68.7	69.2	69.2	70.6	68.9	68.4	70.6	71.6	74.7	73.2	77.5	79.6	81.9	84.4	83.6	
5	125	70.0	69.5	69.1	69.6	70.1	71.6	72.1	73.3	74.1	75.3	76.5	78.1	79.1	81.0	83.6	80.5
6	160	70.0	71.0	71.0	71.3	72.5	72.8	73.1	73.3	74.3	74.8	75.1	76.5	77.8	79.6	81.1	78.0
7	200	69.7	70.6	70.4	70.2	70.1	69.6	69.7	70.1	71.4	72.2	74.1	75.8	78.2	79.4	80.9	76.8
8	250	69.3	65.8	69.6	69.8	69.6	70.1	70.6	71.8	72.8	74.3	75.8	77.4	78.6	80.0	81.0	77.9
9	315	65.0	65.3	70.6	71.3	70.5	71.0	71.1	72.0	73.3	73.8	75.1	76.6	78.5	79.3	79.5	76.2
10	400	68.5	69.5	70.0	71.0	70.5	71.8	70.8	71.3	72.8	73.8	75.1	76.9	78.3	79.1	78.8	74.5
11	500	71.1	72.2	71.4	71.9	72.7	71.2	70.9	71.9	72.9	73.9	74.7	76.0	77.6	77.9	77.6	73.3
12	630	69.6	70.7	71.7	73.2	72.4	71.9	72.4	73.4	73.9	74.9	75.6	77.0	76.7	75.7	75.7	71.8
13	800	74.8	74.0	73.5	77.5	75.2	72.2	73.0	74.4	73.7	75.0	75.4	76.1	77.7	77.0	76.0	72.7
14	1000	75.0	74.6	76.2	75.5	78.5	74.4	74.4	73.0	73.7	74.5	75.7	76.8	78.0	77.9	76.4	72.6
15	1250	75.7	77.2	77.6	78.7	77.0	75.2	73.4	74.4	74.2	75.2	76.9	76.7	77.2	77.1	75.9	72.3
16	1600	75.6	76.1	75.9	75.9	75.1	74.2	73.4	73.0	74.0	75.4	76.5	77.7	78.5	75.5	76.2	70.8
17	2000	86.3	87.4	87.8	88.8	86.1	84.2	81.6	78.9	81.7	80.1	84.6	83.7	83.1	79.1	77.2	74.7
18	2500	77.9	78.9	79.2	75.4	77.9	76.4	74.4	74.9	76.4	76.9	78.9	80.0	79.4	74.9	71.0	71.0
19	3150	77.0	78.6	79.3	80.1	78.1	76.8	75.6	76.3	78.5	79.3	80.8	82.8	82.3	78.1	75.8	72.4
20	4000	80.5	82.1	82.7	83.7	81.9	79.4	80.2	79.4	85.1	85.2	85.4	86.0	86.7	84.6	83.1	77.4
21	5000	78.2	79.7	79.7	80.7	79.5	76.7	76.4	77.2	79.2	81.0	81.0	83.0	84.4	82.0	78.7	74.2
22	6300	77.5	79.2	80.9	83.4	78.7	77.1	76.9	77.1	81.9	81.3	82.3	83.4	83.4	80.6	77.7	73.1
23	8000	77.4	79.2	80.0	81.8	79.0	77.4	77.5	78.4	81.2	83.9	83.4	84.4	84.7	81.5	78.2	73.4
24	10000	77.4	78.8	79.6	81.4	77.9	76.6	76.6	76.1	79.9	81.6	82.6	84.7	84.3	81.8	79.0	73.0
25	12500	76.9	77.7	78.4	80.2	77.7	76.9	75.1	75.6	77.9	79.4	82.3	82.7	80.4	77.7	71.5	71.5
26	16000	75.2	76.9	78.4	79.7	76.5	74.0	74.0	74.2	77.9	79.2	79.7	81.8	81.9	79.2	76.2	70.5
27	20000	75.6	76.9	77.9	78.0	76.5	75.4	73.5	76.5	78.3	79.5	81.1	81.4	79.0	76.2	70.9	

ORIGINAL MICROPHONE RADII, FEET

100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0	100.0
-------	-------	-------	-------	-------	-------	-------	-------	-------	-------	-------	-------	-------	-------	-------	-------	-------	-------



TABLE I. - Continued. SAMPLE OUTPUT FROM WORKING DATA GENERATION SUBROUTINE WODAG

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
 LEWIS RESEARCH CENTER  
 PROPULSION SYSTEMS ACOUSTICS BRANCH  
 SAMPLE NOISE DATA

CONFIGURATION NO 100  
 SPEED = 1800. RPM  
 PERCENT SPEED = 75.0

R E F E R R E D A R R A Y

DATA AT 100.0 FT RADIUS WITH NO ATMOSPHERIC ATTENUATION  
 (FCR POWER AND DIRECTIVITY COMPUTATIONS)

COMPUTED FREQ	R E F E R R E D A R R A Y															
	10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.
1	69.0	66.1	64.9	65.9	68.6	67.6	67.2	68.9	66.4	68.7	69.6	70.6	73.1	74.4	77.9	77.9
2	44.3	65.6	65.1	63.8	65.0	65.0	65.8	66.1	66.3	66.8	69.3	71.4	74.1	76.1	79.0	80.0
3	65.3	65.8	64.0	64.1	64.8	65.3	65.1	66.8	68.5	70.3	72.8	74.0	77.0	79.3	83.0	83.3
4	68.7	69.2	69.2	70.6	68.9	68.4	70.6	71.6	71.9	74.7	75.2	77.5	79.6	81.9	84.4	83.6
5	70.0	69.5	69.1	69.0	70.1	71.6	72.1	73.3	74.1	75.3	76.5	78.1	79.1	81.0	83.6	80.5
6	160	70.0	71.0	71.3	72.5	72.8	73.1	73.3	74.3	74.8	75.1	76.5	77.8	79.6	81.1	78.0
7	69.7	70.6	70.4	70.2	70.1	69.6	69.7	70.1	71.4	72.2	74.1	75.8	78.2	79.4	80.9	76.8
8	69.3	69.8	69.6	69.8	69.6	70.1	70.6	71.8	72.8	74.3	75.8	77.4	78.6	80.0	81.0	77.9
9	65.0	69.3	70.6	71.3	70.5	71.0	71.1	72.0	73.3	73.8	75.1	76.6	78.5	79.3	79.5	76.2
10	68.6	69.6	70.1	71.1	70.6	71.9	70.9	71.4	72.9	73.9	75.2	77.0	78.4	79.2	78.9	74.6
11	71.2	72.3	71.5	72.0	72.8	71.3	71.0	72.0	73.0	74.0	74.8	76.1	77.7	78.0	77.7	73.4
12	69.7	70.8	71.8	73.3	72.5	71.6	72.0	72.5	73.5	74.0	75.0	75.7	77.1	76.8	75.8	71.9
13	74.9	74.1	73.6	77.6	75.3	72.3	73.1	74.5	73.8	75.1	75.5	76.2	77.8	77.1	76.1	72.8
14	75.1	74.7	76.3	75.6	78.6	74.5	74.5	73.1	73.8	74.6	75.8	76.9	78.1	78.0	76.5	72.7
15	1250	75.9	77.4	77.8	78.9	77.2	75.4	75.3	74.4	75.4	77.1	76.9	77.4	77.3	76.1	72.5
16	1600	75.9	76.4	76.2	76.2	75.4	73.7	73.3	74.3	75.7	76.8	78.0	78.8	75.8	74.5	71.1
17	2000	86.7	87.8	88.2	89.2	86.5	84.6	82.0	79.3	82.1	80.5	85.0	84.1	83.5	77.6	75.1
18	2500	78.4	75.4	79.7	79.9	78.4	76.9	74.9	75.4	76.9	77.4	80.5	79.9	77.4	75.4	71.5
19	3150	77.8	75.4	80.1	80.9	78.9	77.6	76.4	77.1	79.3	80.1	81.6	83.1	78.9	76.6	73.2
20	4000	81.6	83.2	83.8	84.8	83.0	80.5	81.3	80.5	86.2	86.3	86.5	87.1	87.9	85.8	78.5
21	5000	79.8	81.3	81.3	82.3	81.1	78.3	78.0	78.8	80.8	81.8	82.6	84.6	86.0	83.6	75.8
22	6300	79.8	81.5	83.2	85.7	81.0	79.4	79.2	84.2	84.2	83.6	84.6	85.7	85.7	80.0	75.4
23	8000	80.6	82.4	83.2	85.0	82.2	80.7	81.6	84.4	87.2	86.7	87.7	88.0	89.8	81.5	76.6
24	10000	82.0	83.3	84.1	85.9	82.4	81.1	81.1	80.6	84.4	86.1	87.1	89.2	88.8	83.6	77.5
25	12500	83.2	84.0	84.7	86.5	84.0	81.4	81.4	81.9	84.2	85.7	88.6	89.0	86.7	84.0	77.8
26	16000	84.0	85.7	87.2	88.5	85.3	82.8	83.0	86.7	88.0	88.5	90.6	90.7	88.0	85.0	79.3
27	20000	87.8	89.1	90.0	90.1	88.7	86.0	87.6	88.6	90.4	91.6	93.2	93.5	91.1	88.4	83.1

TABLE I. - Continued. SAMPLE OUTPUT FROM WORKING DATA GENERATION SUBROUTINE WODAG

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
LEWIS RESEARCH CENTER  
PROPULSION SYSTEMS ACOUSTICS BRANCH  
SAMPLE NOISE DATA

CONFIGURATION NO 100  
SPEED = 1800. RPM  
PERCENT SPEED = 75.0

BAND	FREQUENCY	ACOUSTIC POWER COMPUTATIONS		SIMPLE SOURCE SPL, R = 100.0 FT
		POWER SPECTRUM	NORMALIZED POWER SPECTRUM	
1	50	118.5	-18.9	71.1
2	63	119.1	-18.3	71.7
3	80	122.2	-15.2	74.8
4	100	124.4	-13.0	77.0
5	125	124.0	-13.3	76.7
6	160	123.0	-14.4	75.6
7	200	122.0	-15.4	74.6
8	250	122.8	-14.6	75.4
9	315	122.3	-15.1	75.0
10	400	122.1	-15.3	74.7
11	500	121.7	-15.7	74.4
12	630	121.4	-15.9	74.1
13	800	122.7	-14.7	75.3
14	1000	123.3	-14.1	75.9
15	1250	123.7	-13.6	76.4
16	1600	123.1	-14.2	75.8
17	2000	131.7	-5.7	84.3
18	2500	125.5	-11.9	78.1
19	3150	127.4	-10.0	80.0
20	4000	132.2	-5.2	84.8
21	5000	129.2	-8.1	81.9
22	6300	130.5	-6.9	83.1
23	8000	132.1	-5.3	84.8
24	10000	132.7	-4.7	85.3
25	12500	132.7	-4.6	85.4
26	16000	134.6	-2.8	87.2
27	20000	137.4	0.	90.0
				6.1 OVERALL
				96.2 OVERALL

TOTAL ACOUSTIC POWER

-13

PWL = 143.5 DB RE 10 WATT

W = 22.5 WATTS



TABLE I. - Continued. SAMPLE OUTPUT FROM WORKING DATA GENERATION SUBROUTINE WODAG

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
 LEWIS RESEARCH CENTER  
 PROPULSION SYSTEMS ACOUSTICS BRANCH  
 SAMPLE NOISE DATA

CONFIGURATION NO 100  
 SPEED = 1800 RPM  
 PERCENT SPEED = 75.0

NORMALIZED POWER SPECTRUM

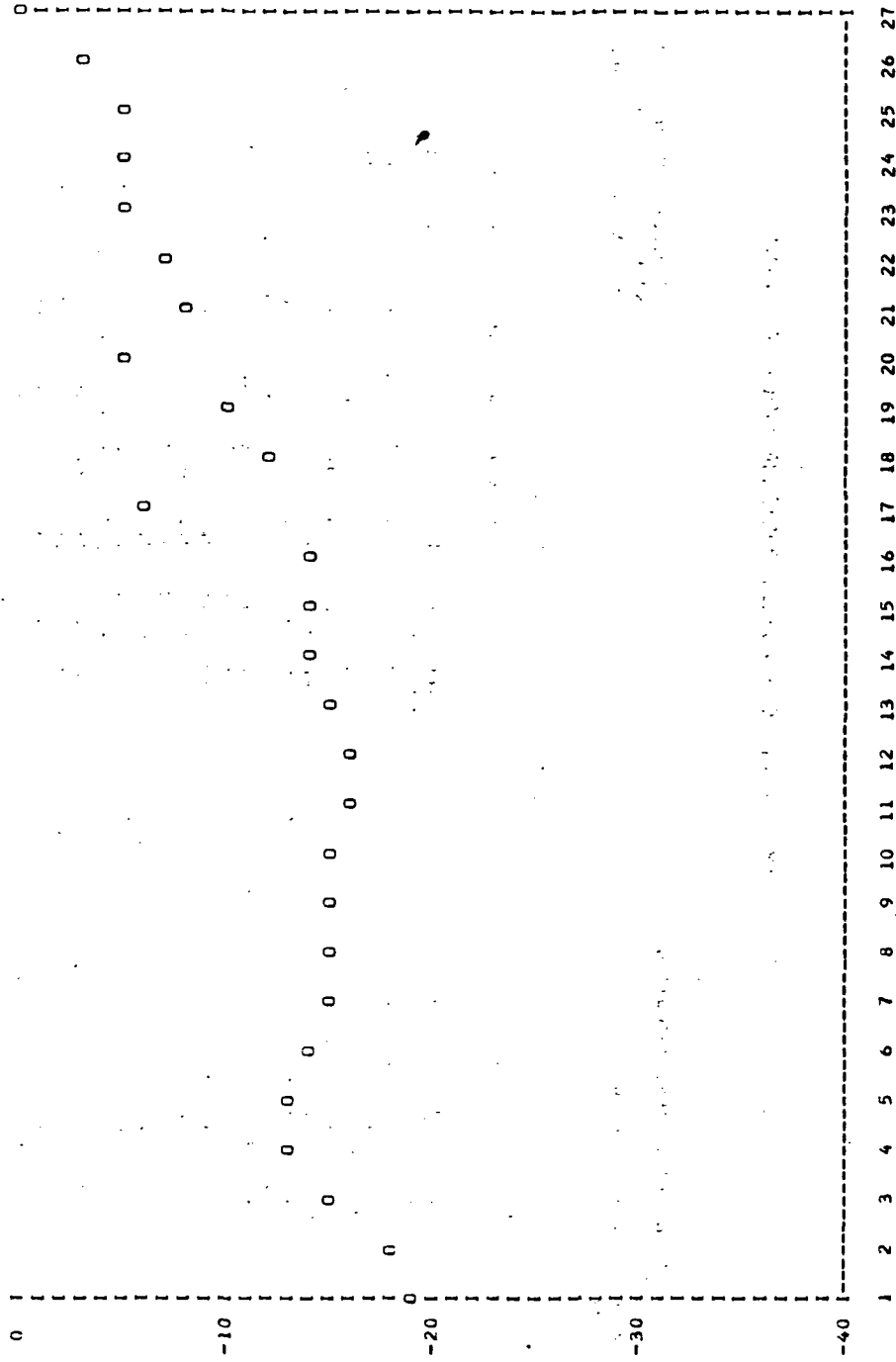


TABLE I. - Continued. SAMPLE OUTPUT FROM WORKING DATA GENERATION SUBROUTINE WODAG

D I R E C T I V I T Y I N D E X

CONFIGURATION NO 100  
SPEED = 1800. RPM  
PERCENT SPEED = 75.0

PANEL FREQUENCY	D I R E C T I V I T Y I N D E X																
	ANGLE: 50.	10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.
1	50	-7.1	-5.0	-6.2	-5.2	-2.5	-3.5	-3.9	-2.2	-4.7	-2.4	-1.5	-0.5	2.0	3.3	6.8	6.8
2	63	-7.4	-6.1	-6.6	-7.9	-6.7	-6.7	-5.9	-5.6	-5.4	-4.9	-2.4	-0.3	2.4	4.4	7.3	8.3
3	80	-5.5	-5.0	-10.8	-10.7	-10.0	-9.5	-9.7	-8.0	-6.3	-4.5	-2.0	-0.8	2.2	4.5	8.2	8.5
4	100	-8.3	-7.5	-7.6	-6.4	-8.1	-8.6	-6.4	-5.4	-5.1	-2.3	-1.8	0.5	2.6	4.9	7.4	6.6
5	125	-6.7	-7.2	-7.6	-7.1	-6.6	-5.1	-4.6	-3.4	-2.6	-1.4	-0.2	1.4	2.4	4.3	6.9	3.8
6	160	-5.6	-4.6	-4.6	-4.3	-3.1	-2.8	-2.5	-2.3	-1.3	-0.8	-0.5	0.9	2.2	4.0	5.5	2.4
7	200	-4.9	-4.0	-4.2	-4.4	-4.5	-5.0	-4.9	-4.5	-3.2	-2.4	-0.5	1.2	3.6	4.8	6.3	2.2
8	250	-6.1	-5.6	-5.8	-5.6	-5.6	-5.3	-4.8	-3.6	-2.6	-1.1	0.4	2.0	3.2	4.6	5.6	2.5
9	315	-5.9	-5.6	-4.3	-3.0	-4.4	-3.9	-3.8	-2.9	-1.6	-1.1	0.2	1.7	3.6	4.4	4.6	1.3
10	400	-6.2	-5.2	-4.7	-3.7	-4.2	-2.9	-3.9	-3.4	-1.9	-0.9	0.4	2.2	3.6	4.4	4.1	-0.2
11	500	-3.2	-2.1	-2.9	-2.4	-1.6	-3.1	-3.4	-2.4	-1.4	-0.4	0.4	1.7	3.3	3.6	3.3	-1.0
12	630	-4.4	-3.3	-2.3	-0.8	-1.6	-2.5	-2.1	-1.6	-0.6	-0.1	0.9	1.6	3.0	2.7	1.7	-2.2
13	800	-0.4	-1.2	-1.7	2.3	0.0	-3.0	-2.2	-0.8	-1.5	-0.2	0.2	0.9	2.5	1.8	0.8	-2.5
14	1000	-0.8	-1.2	0.4	-0.3	2.7	-1.4	-1.4	-2.8	-2.1	-1.3	-0.1	1.0	2.2	2.1	0.6	-3.2
15	1250	-0.5	1.0	1.4	2.5	0.8	-1.0	-1.1	-1.8	-2.0	-1.0	0.7	0.5	1.0	0.9	-0.3	-3.9
16	1600	0.1	0.6	0.4	0.4	-0.4	-1.3	-2.1	-2.5	-1.5	-0.1	1.0	2.2	3.0	-0.0	-1.3	-4.7
17	2000	4.4	3.5	3.9	4.9	2.2	-0.3	-2.3	-5.0	-2.2	-3.8	0.7	-0.2	-0.8	-4.8	-6.7	-9.2
18	2500	0.3	1.3	1.6	1.8	0.3	-1.2	-3.2	-2.7	-1.2	-0.7	1.3	2.4	1.8	-0.7	-2.7	-6.6
19	3150	-2.2	-0.6	0.1	0.9	-1.1	-2.4	-3.6	-2.9	-0.7	0.1	1.6	3.6	3.1	-1.1	-3.4	-6.8
20	4000	-3.2	-1.6	-1.0	0.0	-1.8	-4.3	-3.5	-4.3	1.4	1.5	1.7	2.3	3.0	0.9	-3.6	-6.3
21	5000	-2.1	-0.6	-0.6	0.4	-0.6	-3.6	-3.9	-3.1	-1.1	-0.1	0.7	2.7	4.1	1.7	-1.6	-6.1
22	6300	-3.3	-1.6	0.1	2.6	-2.1	-3.7	-3.9	-3.7	1.1	0.5	1.5	2.6	2.6	-0.2	-3.1	-7.7
23	8000	-4.1	-2.3	-1.5	0.3	-2.5	-4.1	-4.0	-3.1	-0.3	2.4	1.9	2.9	3.2	0.0	-3.3	-8.1
24	10000	-3.4	-2.0	-1.2	0.6	-2.9	-4.2	-4.2	-4.7	-0.9	0.8	1.8	3.9	3.5	1.0	-1.8	-7.8
25	12500	-3.2	-1.4	-0.7	1.1	-1.4	-4.2	-4.0	-3.5	-1.2	0.3	1.3	3.2	3.6	1.3	-1.4	-7.6
26	16000	-3.2	-1.5	0.0	1.3	-1.9	-4.4	-4.4	-4.2	-0.5	0.8	1.3	3.4	3.5	0.8	-2.2	-7.9
27	20000	-2.3	-1.0	0.0	0.1	-1.4	-4.0	-2.4	-4.4	-1.4	0.4	1.6	3.2	3.5	1.1	-1.7	-7.0
OVERALL																	
-2.2 -0.9 -0.1 0.9 -1.3 -3.5 -3.4 -3.9 -0.8 0.4 1.3 2.8 3.1 1.1 -0.4 -3.8																	

TABLE I. - Continued. SAMPLE OUTPUT FROM WORKING DATA GENERATION SUBROUTINE WODAG

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
 LEA'S RESEARCH CENTER  
 PROPULSION SYSTEMS ACOUSTICS BRANCH  
 SAMPLE NOISE DATA

CONFIGURATION NO. 100  
 SPEED = 1800. RPM  
 PERCENT SPEED = 75.0

A T M O S P H E R I C A T T E N U A T I O N

STANDARD DAY EXCESS ATMOSPHERIC ATTENUATION PER THOUSAND FEET

CUMULATED FROM REFERRED ARRAY CONSIDERING SPECTRUM SHAPES

PANE FREQUENCY	ANGLE																
	10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.	
1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1
2	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1
3	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1
4	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1	0.1
5	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2
6	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2	0.2
7	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3	0.3
8	0.4	0.4	0.4	0.4	0.4	0.4	0.4	0.4	0.4	0.4	0.4	0.4	0.4	0.4	0.4	0.4	0.4
9	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5
10	0.6	0.6	0.6	0.6	0.6	0.6	0.6	0.6	0.6	0.6	0.6	0.6	0.6	0.6	0.6	0.6	0.6
11	0.7	0.7	0.7	0.7	0.7	0.7	0.7	0.7	0.7	0.7	0.7	0.7	0.7	0.7	0.7	0.7	0.7
12	0.9	0.9	0.9	0.9	0.9	0.9	0.9	0.9	0.9	0.9	0.9	0.9	0.9	0.9	0.9	0.9	0.9
13	1.2	1.2	1.2	1.2	1.2	1.2	1.2	1.2	1.2	1.2	1.2	1.2	1.2	1.2	1.2	1.2	1.2
14	1.5	1.5	1.5	1.5	1.5	1.5	1.5	1.5	1.5	1.5	1.5	1.5	1.5	1.5	1.5	1.5	1.5
15	1.9	1.9	1.9	1.9	1.9	1.9	1.9	1.9	1.9	1.9	1.9	1.9	1.9	1.9	1.9	1.9	1.9
16	2.5	2.5	2.5	2.5	2.5	2.5	2.5	2.5	2.5	2.5	2.5	2.5	2.5	2.5	2.5	2.5	2.5
17	3.1	3.1	3.1	3.1	3.1	3.1	3.1	3.1	3.1	3.1	3.1	3.1	3.1	3.1	3.1	3.1	3.1
18	4.0	4.0	4.0	4.0	4.0	4.0	4.0	4.0	4.0	4.0	4.0	4.0	4.0	4.0	4.0	4.0	4.0
19	5.7	5.7	5.7	5.7	5.7	5.7	5.7	5.7	5.7	5.7	5.7	5.7	5.7	5.7	5.7	5.7	5.7
20	7.8	7.8	7.8	7.8	7.8	7.8	7.8	7.8	7.8	7.8	7.8	7.8	7.8	7.8	7.8	7.8	7.8
21	10.9	10.9	10.9	10.9	10.9	10.9	10.9	10.9	10.9	10.9	10.9	10.9	10.9	10.9	10.9	10.9	10.9
22	15.7	15.7	15.7	15.7	15.7	15.7	15.7	15.7	15.7	15.7	15.7	15.7	15.7	15.7	15.7	15.7	15.7
23	22.7	22.7	22.7	22.7	22.7	22.7	22.7	22.7	22.7	22.7	22.7	22.7	22.7	22.7	22.7	22.7	22.7
24	32.4	32.4	32.4	32.4	32.4	32.4	32.4	32.4	32.4	32.4	32.4	32.4	32.4	32.4	32.4	32.4	32.4
25	46.3	46.3	46.3	46.3	46.3	46.3	46.3	46.3	46.3	46.3	46.3	46.3	46.3	46.3	46.3	46.3	46.3
26	66.0	66.0	66.0	66.0	66.0	66.0	66.0	66.0	66.0	66.0	66.0	66.0	66.0	66.0	66.0	66.0	66.0
27	94.2	94.2	94.2	94.2	94.2	94.2	94.2	94.2	94.2	94.2	94.2	94.2	94.2	94.2	94.2	94.2	94.2



TABLE I. - Continued. SAMPLE OUTPUT FROM WORKING DATA GENERATION SUBROUTINE WODAG

ATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
LEWIS RESEARCH CENTER  
PROPULSION SYSTEMS ACCLUSTLS BRANCH  
SAMPLE NOISE DATA

CONFIGURATION NO 100  
SPEED = 1800. RPM  
PERCENT SPEED = 75.0

S T A N D A R D D A Y D A T A

DATA AT 100.0 FT RADIUS ON 59F, 70PC RH DAY

COMPLETED CASPL	ANGLF	10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.
1	69.0	65.1	64.9	65.9	68.6	67.6	67.2	68.9	68.4	68.7	69.6	70.6	73.1	74.4	77.9	77.9	77.9
2	64.3	65.6	65.1	63.8	65.0	65.0	65.8	66.1	66.3	66.8	69.3	71.4	74.1	76.1	79.0	80.0	80.0
3	65.3	65.8	64.0	64.1	64.8	65.3	65.1	66.8	68.5	70.3	72.8	74.0	77.0	79.3	83.0	83.3	83.3
4	68.7	65.2	65.2	70.0	68.9	68.4	70.6	71.0	71.9	74.7	75.2	77.5	79.6	81.9	84.4	83.6	83.6
5	70.3	65.5	69.1	69.6	70.1	71.6	72.1	73.3	74.1	75.3	76.5	78.1	79.1	81.0	83.6	80.5	80.5
6	70.0	71.0	71.0	71.3	72.5	72.8	73.1	73.3	74.3	74.8	75.1	76.5	77.8	79.6	81.1	78.0	78.0
7	65.7	70.0	70.4	70.2	70.1	69.6	69.7	70.1	71.4	72.2	74.1	75.8	78.2	79.4	80.9	76.8	76.8
8	65.5	65.8	64.6	69.8	69.6	70.1	70.6	71.8	72.8	74.3	75.8	77.4	78.6	80.0	81.0	77.9	77.9
9	31.5	69.3	70.6	71.3	70.5	71.0	71.1	72.0	73.3	73.8	75.1	76.6	78.5	79.3	75.5	76.2	76.2
10	68.5	65.5	70.0	71.0	70.5	71.8	70.8	71.3	72.8	73.8	75.1	76.9	78.3	79.1	78.8	74.5	74.5
11	50.0	71.1	72.2	71.4	71.9	72.7	71.2	70.9	71.9	72.9	74.7	76.0	77.6	77.9	77.6	73.3	73.3
12	64.0	69.6	70.7	71.7	73.2	72.4	71.5	71.9	72.4	73.4	74.9	75.6	76.7	77.0	75.7	71.8	71.8
13	80.0	74.8	74.0	73.5	77.5	75.2	72.2	73.0	74.4	73.7	75.0	75.4	76.1	77.7	77.0	76.0	72.7
14	100.0	75.0	74.6	76.2	75.5	78.5	74.4	74.4	73.0	73.7	74.5	75.7	76.8	78.0	77.9	76.4	72.6
15	125.0	75.7	77.7	77.6	78.7	77.0	75.1	74.4	74.4	74.2	75.2	76.9	76.7	77.1	75.9	72.3	72.3
16	160.0	75.0	76.1	75.9	75.9	75.1	74.2	73.4	73.0	74.0	75.4	76.5	77.7	78.5	75.5	74.2	70.8
17	200.0	80.4	81.5	87.9	88.9	86.2	84.3	81.7	79.0	81.8	80.2	84.7	83.8	83.2	79.2	77.3	74.8
18	250.0	78.0	79.0	79.3	79.5	78.0	76.5	74.5	75.0	76.5	77.0	79.0	80.1	79.5	77.0	75.0	71.1
19	315.0	77.2	76.8	79.5	80.3	78.3	77.0	75.8	76.5	78.7	79.5	81.0	83.0	82.5	78.3	76.0	72.6
20	400.0	80.9	82.5	83.1	84.1	82.3	79.8	80.6	79.8	85.5	85.6	85.8	86.4	87.1	85.0	80.5	77.8
21	500.0	78.7	80.2	80.2	81.2	80.0	77.2	76.9	77.7	79.7	80.7	81.5	83.5	84.9	82.5	79.2	74.7
22	630.0	78.2	79.5	81.6	84.1	79.4	77.8	77.6	77.8	82.6	82.0	83.0	84.1	84.1	81.3	78.4	73.8
23	800.0	78.4	80.2	81.0	82.8	80.0	78.4	78.5	79.4	82.2	84.9	84.4	85.4	85.7	82.5	79.2	74.4
24	1000.0	78.7	80.1	80.9	82.7	79.2	77.9	77.4	77.4	81.2	82.9	83.9	86.0	85.6	83.1	80.3	74.3
25	1250.0	78.6	79.4	80.1	81.9	79.4	78.0	76.8	77.3	79.6	81.1	82.1	84.0	84.4	82.1	79.4	73.2
26	1600.0	77.4	75.1	80.6	81.9	78.7	76.2	76.4	80.1	81.4	81.4	81.9	84.0	84.1	81.4	78.4	72.7
27	2000.0	78.3	79.7	80.7	80.8	79.2	78.1	78.1	76.3	79.3	81.1	82.3	83.9	84.2	81.8	78.9	73.6

PERCEIVED NOISE ON 100.0 FT RADIUS, PNOB

ANGLF	10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.
	104.6	105.7	106.1	107.2	105.0	103.3	102.8	102.5	106.2	106.7	107.5	108.4	109.1	107.3	104.9	101.7



TABLE I. - Continued. SAMPLE OUTPUT FROM WORKING DATA GENERATION SUBROUTINE WODAG

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
 LEWIS RESEARCH CENTER  
 PROPELLION SYSTEMS ACUSTICS BRANCH  
 SAMPLE NOISE DATA

CONFIGURATION NO. 100  
 SPEED = 1800. RPM  
 PERCENT SPEED = 75.0

COMPLETED CASE#	SIDELINE EXTRAPOLATED DATA																
	ANGLE	10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.
		STANDARD DAY DATA ON 370. FT SIDELINE, INCORPORATING EXCESS ATMOSPHERIC ATTENUATION															
FAN#	FREQUENCY	57.5	60.4	71.7	75.5	75.2	75.3	75.5	78.1	78.5	79.3	79.3	78.7	76.5	74.9	69.5	
1	50	42.3	45.3	47.5	50.7	54.9	55.0	55.3	57.4	55.0	57.2	57.7	58.0	59.4	59.2	63.5	57.1
2	63	37.5	44.8	47.7	48.0	51.3	52.4	53.9	54.0	54.9	55.3	57.4	58.8	60.4	60.9	61.6	59.2
3	80	38.5	45.0	48.5	48.8	51.1	52.6	53.2	55.3	57.1	58.8	60.9	61.3	63.3	64.0	65.5	62.5
4	100	41.8	48.4	51.7	52.3	55.2	55.7	58.7	60.1	61.8	62.7	63.2	64.8	65.9	66.6	66.9	62.8
5	125	43.1	48.6	51.6	54.3	58.3	58.9	60.1	61.8	62.7	63.8	64.5	65.4	65.4	65.7	66.1	59.6
6	160	44.0	50.1	53.5	56.0	58.7	60.1	61.1	61.7	62.9	63.2	63.1	63.8	64.0	64.3	63.6	57.1
7	200	42.5	49.8	52.8	54.9	58.3	58.9	57.7	58.5	60.0	60.6	62.1	63.1	64.4	64.1	63.3	55.8
8	250	42.3	48.8	52.0	54.4	55.8	57.4	58.6	60.2	61.3	62.7	63.8	64.7	64.8	64.6	63.4	56.9
9	315	41.5	48.2	52.9	52.9	56.6	58.2	59.1	60.4	61.8	62.2	63.1	63.8	64.6	63.9	61.8	55.1
10	400	40.7	48.2	52.2	55.5	58.6	59.0	58.7	59.8	61.3	62.1	63.0	64.1	64.4	63.6	61.0	53.2
11	500	43.0	50.8	53.5	56.3	58.7	58.3	58.8	59.7	60.6	61.8	62.2	62.6	63.1	63.6	62.3	59.7
12	630	41.1	49.1	53.7	57.5	58.4	58.6	59.7	60.6	61.8	62.1	62.7	63.0	61.0	57.7	50.2	
13	800	45.8	52.1	55.3	61.7	61.1	59.2	60.7	62.6	62.0	63.2	63.1	63.1	63.6	61.2	57.9	
14	1000	45.4	52.4	57.5	59.6	64.2	61.3	62.1	61.1	61.9	62.6	63.4	63.7	63.8	62.0	58.1	
15	1250	45.3	54.7	59.0	62.0	62.0	62.0	62.7	62.4	62.3	63.2	64.5	63.5	62.8	61.0	57.3	
16	1600	44.0	53.0	57.0	60.5	60.5	60.8	60.8	60.9	62.0	63.3	63.9	64.3	63.9	59.2	55.3	
17	2000	53.4	63.7	66.5	72.2	71.3	73.6	68.9	66.6	69.6	67.8	71.9	70.1	68.3	62.5	57.9	
18	2500	43.3	54.4	59.4	62.4	62.8	62.6	61.4	62.4	64.1	64.4	65.9	66.2	64.3	59.9	55.0	
19	3150	35.1	52.6	58.5	62.4	62.5	62.6	62.2	63.5	65.8	66.4	67.4	68.5	66.6	60.4	55.0	
20	4000	38.4	54.1	60.7	65.2	65.6	64.6	66.4	66.1	72.0	71.9	71.6	71.2	70.4	66.0	58.1	
21	5000	30.1	48.9	55.8	60.8	62.2	61.0	61.8	63.2	65.4	66.2	66.4	67.3	67.1	62.2	54.9	
22	6300	15.8	43.8	54.1	61.4	59.7	60.0	61.1	62.0	67.0	66.1	66.5	66.3	64.4	58.6	51.0	
23	8000	5.7	37.2	49.2	56.9	57.6	58.4	59.9	61.6	64.7	67.1	65.8	65.3	63.3	56.4	47.2	
24	10000	0.	27.6	42.7	52.1	53.1	54.7	56.5	57.0	61.1	62.6	62.5	62.8	59.5	52.4	42.0	
25	12500	0.	13.2	33.0	44.6	47.9	48.9	51.3	53.0	55.7	56.8	56.6	56.3	53.0	44.9	32.4	
26	16000	0.	0.	20.8	35.3	39.7	41.9	44.8	46.7	50.9	51.6	50.5	49.7	45.1	34.8	18.8	
27	20000	0.	0.	3.3	21.3	24.6	33.3	33.4	38.9	42.7	43.8	42.8	40.6	34.6	21.9	1.4	

ANGLE	INVERSE SQUARE LAW ATTENUATION FROM 100. FT RADIUS TO 370. FT SIDELINE																
	10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.	
	PERCEIVED NOISE ON 370. FT SIDELINE, PNDB																
20.6	20.7	17.4	15.2	13.7	12.6	11.9	11.5	11.4	11.5	11.4	11.5	11.9	12.6	13.7	15.2	17.4	20.7
65.3	80.1	85.3	85.3	89.2	88.9	88.7	88.9	92.7	93.0	93.3	93.3	92.5	88.7	83.2	74.8		

TABLE I. - Concluded. SAMPLE OUTPUT FROM WORKING DATA GENERATION SUBROUTINE WODAG

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
LEWIS RESEARCH CENTER  
PROPULSION SYSTEMS ACOUSTICS BRANCH  
SAMPLE NOISE DATA

CONFIGURATION NU 100  
SPEED = 1800. PPM  
PERCENT SPEED = 75.0

S I D E L I N E   E X T R A P O L A T E D   D A T A

FAN FREQUENCY	STANDARD-LAY DATA ON 1000. FT SIDELINE, INCORPORATING EXCESS ATMOSPHERIC ATTENUATION																
	ANGLE 10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.	
1	44.1	53.5	59.4	63.0	64.3	64.3	64.4	64.8	66.6	67.0	68.0	68.1	67.8	66.5	65.5	60.2	
2	33.4	36.0	38.7	42.0	46.2	48.3	46.6	48.7	46.3	48.5	49.0	49.3	50.7	50.5	51.7	48.4	
3	28.6	36.0	38.5	35.8	42.6	43.7	45.2	45.9	46.2	46.6	48.7	50.1	51.7	52.1	52.8	50.4	
4	25.4	36.2	37.8	43.1	42.3	43.9	44.4	46.6	48.4	50.1	52.1	52.6	54.5	55.3	56.8	53.7	
5	32.7	39.5	42.9	46.5	46.4	47.0	49.9	51.3	51.8	54.4	54.5	56.1	57.1	57.8	58.1	53.9	
6	33.8	39.7	42.7	45.5	47.6	50.2	51.4	53.0	53.9	55.0	55.8	56.7	56.6	56.9	57.2	50.7	
7	33.5	41.3	44.5	47.1	49.9	51.3	52.3	53.0	54.1	54.5	54.3	55.0	55.2	55.4	54.6	48.0	
8	32.9	40.5	43.8	45.9	47.4	48.0	48.9	49.7	51.1	51.8	53.3	54.2	55.5	55.1	54.3	46.7	
9	32.0	39.5	42.9	45.4	46.8	48.5	49.7	51.3	52.5	53.8	54.9	55.8	55.8	55.6	54.3	47.6	
10	31.7	38.7	43.7	46.8	47.6	49.3	50.1	51.4	52.9	53.2	54.1	54.9	55.6	54.8	52.6	45.6	
11	30.0	38.5	42.9	46.3	47.5	49.9	49.7	50.6	52.3	53.1	54.0	55.0	55.3	54.4	51.7	43.6	
12	31.7	40.8	44.0	47.0	49.5	49.2	49.6	51.1	52.2	53.1	53.4	54.0	54.4	53.0	50.2	41.9	
13	29.1	38.7	43.9	48.5	49.0	49.3	50.5	51.4	52.6	52.9	53.5	53.4	53.6	51.5	47.9	39.9	
14	32.9	41.3	45.2	51.9	51.4	49.7	51.3	53.2	52.6	53.8	53.7	53.6	54.0	51.4	47.7	40.1	
15	31.4	41.1	47.3	49.5	54.4	51.6	52.4	51.5	52.4	53.0	53.7	54.0	53.9	51.9	47.6	39.1	
16	29.8	42.6	48.0	52.1	52.4	52.8	52.8	52.5	53.3	54.6	53.5	53.5	52.6	50.6	46.3	37.7	
17	26.4	39.8	45.2	48.5	49.8	50.4	50.5	50.7	51.8	53.1	53.6	53.9	53.3	48.2	43.7	34.7	
18	33.5	45.3	55.9	60.5	60.1	59.7	58.1	56.0	59.0	57.2	61.1	59.2	57.1	50.8	45.3	36.7	
19	38.4	45.7	49.8	50.8	51.0	50.1	50.1	51.1	52.8	53.1	54.5	54.5	52.2	47.1	41.1	30.1	
20	33.4	42.7	48.2	49.1	49.8	49.8	49.7	51.2	53.5	54.1	54.9	55.7	53.2	46.0	39.1	27.0	
21	31.1	42.2	48.5	50.5	50.3	52.5	58.5	58.5	58.3	57.7	56.9	55.3	49.6	39.5	26.3	0.	
22	33.5	41.4	46.6	44.5	45.9	47.6	50.0	50.7	50.8	50.8	49.5	42.9	32.5	14.9	0.	0.	
23	25.6	37.2	38.2	39.9	41.8	43.2	48.3	47.2	47.1	48.2	42.8	34.6	22.5	0.2	0.	0.	
24	12.1	26.2	30.4	33.2	36.1	38.5	41.9	43.9	41.9	40.0	35.9	25.4	9.8	0.	0.	0.	
25	0.	0.	0.	11.6	17.8	22.6	26.2	27.8	32.2	33.4	32.3	30.6	24.2	11.9	0.	0.	
26	0.	0.	0.	0.	1.1	6.6	11.7	14.7	17.9	18.7	17.1	14.1	6.4	0.	0.	0.	
27	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	
28	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	

I N V E R S E   S Q U A R E   L A W   A T T E N U A T I O N   F R O M   1 0 0 . F T   R A D I U S   T O   1 0 0 0 . F T   S I D E L I N E

ANGLE	10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.
35.2	29.3	26.0	23.8	22.3	21.2	20.5	20.1	20.0	20.1	20.5	21.2	22.3	23.8	26.0	29.3	
P E R C E I V E L   N O I S E   O N   1 0 0 0 . F T   S I D E L I N E ,   P N D B																
45.5	65.0	71.7	76.4	76.9	76.9	76.6	76.2	79.7	79.9	83.3	79.8	78.6	74.3	69.2	63.3	



TABLE II. - SAMPLE OUTPUT FROM SUBROUTINE TABLE

FREQUENCY	DATA ADJUSTED TO STANDARD DAY CF 15 DEGREES C, 70 PERCENT RELATIVE HUMIDITY																	SIMPLE SOURCE SPL	POWER LEVEL (PWL)
	SPL RE .00002 N/SQ M PHL RE .1 PICOWATT																		
	ANGLE, DEG																		
	10	20	30	40	50	60	70	80	90	100	110	120	130	140	150	160			
	1/3-OCTAVE BAND SOUND PRESSURE LEVELS (SPL) CN 30.5 METER RADII																		
50	69.0	66.1	64.9	65.9	68.6	67.6	67.2	68.9	66.4	68.7	69.6	70.6	73.1	74.4	77.9	77.9	71.1	118.5	
63	64.3	65.6	65.1	63.8	65.0	65.0	65.8	66.1	66.3	66.8	69.3	71.4	74.1	76.1	79.0	80.0	71.7	119.1	
80	65.3	65.8	64.0	64.1	64.8	65.3	65.1	66.8	68.5	70.3	72.8	74.0	77.0	75.3	83.0	83.3	74.8	122.2	
100	68.7	65.2	69.2	70.6	68.9	68.4	70.6	71.6	71.9	74.7	75.2	77.5	79.6	81.9	84.4	83.6	77.0	124.4	
125	70.0	69.5	69.1	69.6	70.1	71.6	72.1	73.3	74.1	75.3	76.5	78.1	79.1	81.0	83.6	80.5	76.7	124.1	
160	70.0	71.0	71.0	71.3	72.5	72.8	73.1	73.3	74.3	74.8	75.1	76.5	77.8	75.6	81.1	78.0	75.6	123.0	
200	69.7	70.6	70.4	70.2	70.1	69.6	69.7	70.1	71.4	72.2	74.1	75.8	78.2	75.4	80.9	76.8	74.6	122.0	
250	69.3	69.8	69.6	68.8	68.6	70.1	70.6	71.8	72.8	74.3	75.8	77.4	78.6	80.0	81.0	77.9	75.4	122.8	
315	69.0	69.3	70.6	71.3	70.5	71.0	71.1	72.0	73.3	73.8	75.1	76.6	78.5	79.3	79.5	76.2	74.9	122.3	
400	68.4	68.4	69.9	70.9	70.4	71.7	70.7	71.2	72.7	73.7	75.0	76.8	78.2	75.0	78.7	74.4	74.7	122.1	
500	71.0	72.1	71.3	71.8	72.6	71.1	70.8	71.8	72.8	73.8	74.6	75.5	77.5	77.8	77.5	73.2	74.3	121.7	
630	68.6	70.7	71.7	73.2	72.4	71.5	71.9	72.4	73.4	73.9	74.9	75.6	77.0	76.7	75.7	71.8	74.1	121.5	
800	74.8	74.0	73.5	77.5	75.2	72.2	73.0	74.4	73.7	75.0	75.4	76.1	77.7	77.0	76.0	72.7	75.3	122.7	
1000	75.0	74.6	76.1	75.5	78.5	74.4	74.4	73.0	73.7	74.5	75.7	76.8	78.0	77.9	76.4	72.6	75.9	123.3	
1250	75.7	77.2	77.6	78.7	77.0	75.2	75.1	74.4	74.2	75.2	76.9	76.7	77.2	77.1	75.9	72.3	76.4	123.8	
1600	75.7	76.2	76.0	76.0	75.2	74.3	73.5	73.1	74.1	75.5	76.6	77.8	78.6	75.6	74.3	70.9	75.8	123.2	
2000	86.4	87.5	87.9	88.9	86.2	84.3	81.7	79.0	81.8	80.2	84.7	83.8	83.2	75.2	77.3	74.8	84.3	131.7	
2500	78.0	79.0	79.3	79.5	78.0	76.5	74.5	75.0	76.5	77.0	79.0	80.1	79.5	77.0	75.0	71.1	78.1	125.5	
3150	77.2	78.8	79.5	80.3	78.3	77.0	75.8	76.5	78.7	79.5	81.0	83.0	82.5	78.3	76.0	72.6	80.0	127.4	
4000	80.8	82.4	83.0	84.0	82.2	79.7	80.5	79.7	85.4	85.5	85.7	86.3	87.0	84.9	80.4	77.7	84.8	132.2	
5000	78.7	80.2	80.2	81.2	80.0	77.2	76.9	77.7	79.7	80.7	81.5	83.5	84.9	82.5	79.2	74.7	81.9	129.3	
6300	78.2	79.9	81.6	84.1	79.4	77.8	77.6	77.8	82.6	82.0	83.0	84.1	84.1	81.3	78.4	73.8	83.1	130.5	
8000	78.3	80.1	80.9	82.8	75.9	78.3	78.4	79.3	82.2	84.8	84.3	85.3	85.6	82.4	79.1	74.3	84.7	132.1	
10000	78.6	80.1	80.9	82.7	75.2	77.9	77.9	77.4	81.2	82.9	83.9	86.0	85.6	82.0	80.2	74.3	85.3	132.7	
12500	78.6	79.4	80.1	81.5	75.4	76.6	76.8	77.3	79.6	81.1	82.1	84.0	84.4	82.1	79.4	73.2	85.4	132.8	
16000	77.4	79.1	80.6	81.9	78.7	76.2	76.2	76.4	80.1	81.4	81.9	84.0	84.1	81.4	78.4	72.7	87.2	134.6	
20000	78.3	79.6	80.6	80.8	79.2	76.6	78.1	76.2	79.3	81.0	82.2	83.8	84.1	81.7	78.9	73.6	90.0	137.4	
OVERALL	91.4	92.6	93.2	94.4	92.1	90.3	89.8	89.6	92.5	93.3	94.3	95.6	96.0	94.5	93.9	91.3	96.2	143.5	
SIDELINE PERCEIVED NOISE LEVELS																			
CISTANCE																			
112.8 M	69.3	80.1	85.3	89.3	89.2	88.9	88.6	88.9	92.7	93.0	93.3	93.2	92.4	88.6	83.1	74.8			
304.8 M	49.5	65.0	71.7	76.4	76.9	76.6	76.2	79.7	79.7	79.9	80.3	79.8	78.6	74.3	69.2	60.3			

TABLE III. - SAMPLE OUTPUT FROM SUBROUTINE DADIFF

## C O M P A R I S O N O F T W O D A T A S E T S

DATA SET ONE	CONFIGURATION	PERCENT SPEED	RPM
NATIONAL AERONAUTICS AND SPACE ADMINISTRATION	10C	75.0	1800.
LEWIS RESEARCH CENTER	27 - 1/2 OCTAVE BANDS FROM	50 TO 20000	HERTZ
PROPULSION SYSTEMS ACOUSTICS BRANCH	16 ANGLES EVERY 10. DEGREES FROM	10. TO 160.	
SAMPLE NOISE DATA			
DATA SET TWO	101	50.0	1200.
NATIONAL AERONAUTICS AND SPACE ADMINISTRATION	27 - 1/2 OCTAVE BANDS FROM	50 TO 20000	HERTZ
LEWIS RESEARCH CENTER	16 ANGLES EVERY 10. DEGREES FROM	10. TO 160.	
PROPULSION SYSTEMS ACOUSTICS BRANCH			
SAMPLE NOISE DATA, SECOND SET			

## P O W E R L E V E L D I F F E R E N C E S

(DATA SET TWO MINUS DATA SET ONE)

BAND FREQUENCY	TOTAL POWER				FRONT QUADRANT				REAR QUADRANT			
	SET TWO	SET ONE	DELTA PWL	OVERALL	SET TWO	SET ONE	DELTA PWL	OVERALL	SET TWO	SET ONE	DELTA PWL	OVERALL
OVERALL	133.5	143.6	-10.0		131.0	136.5	-7.9		130.0	141.7	-11.8	
BAND FREQUENCY												
1	50	106.2	118.5	-12.3	100.0	112.0	-11.5	105.0	117.4	-12.4		
2	63	106.4	119.1	-12.7	99.1	105.6	-10.7	105.5	118.5	-13.0		
3	80	108.5	122.2	-13.7	99.3	110.2	-10.9	107.9	121.9	-14.0		
4	100	110.6	124.4	-13.8	102.5	114.7	-12.2	109.9	123.9	-14.0		
5	125	111.2	124.1	-12.5	105.2	116.1	-10.9	109.9	123.3	-13.4		
6	160	110.1	123.0	-12.9	105.4	117.1	-11.7	108.3	121.7	-13.4		
7	200	108.7	122.0	-13.3	103.1	114.6	-11.5	107.3	121.1	-13.8		
8	250	109.9	122.8	-12.9	104.3	115.1	-10.8	108.5	122.0	-13.5		
9	315	109.7	122.3	-12.6	105.2	115.7	-10.5	107.8	121.2	-13.4		
10	400	110.0	122.1	-12.1	105.6	115.5	-9.9	108.0	121.0	-13.0		
11	500	110.4	121.7	-11.3	106.6	116.2	-9.6	108.0	120.2	-12.2		
12	630	111.0	121.5	-10.5	107.8	116.7	-8.9	108.2	119.7	-11.6		
13	800	113.6	122.7	-9.1	110.7	115.0	-8.2	110.4	120.3	-9.9		
14	1000	117.0	123.3	-6.3	114.0	119.5	-5.9	114.0	120.6	-6.6		
15	1250	120.3	123.8	-3.5	118.0	120.5	-2.5	116.5	120.7	-4.2		
16	1600	121.0	123.2	-2.2	118.5	115.4	-0.8	117.4	120.9	-3.5		
17	2000	121.2	131.7	-10.5	117.9	120.0	-12.1	118.5	126.8	-8.3		
18	2500	123.2	125.5	-2.3	120.0	122.1	-2.2	120.4	122.8	-2.4		
19	3150	124.2	127.4	-3.2	121.0	123.1	-2.0	121.3	125.4	-6.1		
20	4000	124.5	132.2	-7.7	121.8	127.3	-5.5	121.1	130.5	-9.3		
21	5000	123.3	129.3	-6.0	120.5	124.6	-4.1	120.1	127.5	-7.4		
22	6300	122.3	130.5	-8.2	120.1	126.3	-6.2	118.2	128.4	-10.2		
23	8000	122.0	132.1	-10.1	120.0	126.5	-6.9	117.8	130.5	-12.8		
24	10000	120.5	132.7	-11.8	119.2	127.3	-8.1	115.9	131.2	-15.3		
25	12500	120.1	132.8	-12.7	118.8	127.5	-9.2	114.3	131.1	-16.8		
26	16000	119.4	134.6	-15.2	118.2	125.7	-11.5	113.2	132.9	-19.7		
27	20000	119.4	137.4	-18.0	118.4	122.6	-14.2	112.4	135.7	-23.2		

TABLE III. - Continued. SAMPLE OUTPUT FROM SUBROUTINE DADIFF

C C C P A R I S C N O F T W O D A T A S E T S

DATA SET ONE  
 NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
 LEWIS RESEARCH CENTER  
 PREPULSION SYSTEMS ACOUSTICS BRANCH  
 SAMPLE NOISE DATA  
 RPM 1800.  
 75.0  
 50 TO 20000 HERTZ  
 10. TO 160.  
 CONFIGURATION  
 10C  
 27 - 1/2 OCTAVE BANDS FROM  
 16 ANGLES EVERY 10. DEGREES FROM

DATA SET TWO  
 NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
 LEWIS RESEARCH CENTER  
 PREPULSION SYSTEMS ACOUSTICS BRANCH  
 SAMPLE NOISE DATA, SECOND SET  
 RPM 1200.  
 50.C  
 50 TO 20000 HERTZ  
 10. TO 160.  
 CONFIGURATION  
 101  
 27 - 1/3 OCTAVE BANDS FROM  
 16 ANGLES EVERY 10. DEGREES FROM

D I F F E R E N C E S O F R E F E R E N C E D A T A

(DATA SET TWO MINUS DATA SET ONE)

BANC FREQUENCY	10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.
OVERALL	-6.2	-5.6	-6.5	-6.6	-7.3	-9.1	-11.4	-11.5	-12.5	-12.5	-12.3	-11.5	-10.5	-12.4	-12.8	
1	-13.1	-12.4	-10.5	-11.2	-13.6	-11.7	-11.5	-12.4	-10.2	-11.0	-11.5	-5.6	-11.4	-13.2	-12.7	-14.2
2	-9.3	-10.6	-11.1	-10.1	-10.8	-10.3	-11.3	-10.9	-10.4	-9.1	-11.2	-11.0	-11.5	-13.2	-14.6	-15.1
3	-10.3	-10.6	-11.1	-10.7	-10.8	-10.6	-11.1	-10.8	-11.3	-11.6	-12.4	-11.2	-12.8	-13.6	-15.5	-15.9
4	-11.2	-5.7	-12.5	-14.3	-12.4	-11.7	-12.9	-12.6	-10.7	-12.5	-12.2	-12.6	-13.4	-15.1	-15.1	-14.9
5	-10.2	-8.7	-8.8	-10.1	-5.8	-11.1	-10.8	-12.2	-11.5	-11.5	-11.5	-12.5	-13.1	-14.9	-15.0	-13.2
6	-10.5	-9.8	-10.2	-11.5	-12.7	-11.8	-11.6	-11.8	-12.3	-12.1	-11.5	-12.6	-13.1	-14.8	-15.1	-14.1
7	-9.4	-5.8	-11.4	-11.4	-12.3	-11.6	-11.4	-11.9	-12.2	-12.0	-12.4	-12.7	-13.7	-15.1	-14.7	-13.3
8	-9.0	-8.2	-9.4	-10.1	-11.3	-11.8	-11.6	-11.6	-11.8	-12.1	-12.0	-12.6	-13.2	-14.7	-14.5	-14.3
9	-7.8	-7.1	-8.7	-10.3	-10.1	-11.5	-10.9	-11.5	-11.9	-12.4	-12.7	-12.6	-13.2	-14.4	-14.5	-14.3
10	-5.5	-5.6	-7.0	-8.8	-9.5	-12.1	-11.1	-11.6	-11.8	-11.9	-11.5	-12.4	-13.2	-14.4	-14.5	-14.3
11	-6.3	-7.1	-7.3	-8.5	-10.6	-10.4	-11.1	-10.8	-11.6	-11.4	-11.4	-11.5	-12.3	-13.1	-13.7	-12.8
12	-3.0	-4.0	-5.6	-8.3	-5.3	-10.3	-11.2	-11.8	-11.5	-10.4	-11.2	-11.6	-11.6	-12.2	-12.5	-12.4
13	-4.7	-4.3	-4.3	-6.4	-5.5	-8.5	-9.3	-11.2	-9.5	-9.8	-9.7	-5.6	-5.6	-10.1	-11.3	-11.6
14	-2.3	-1.0	-3.5	-4.1	-8.9	-7.8	-8.5	-7.3	-7.1	-6.1	-6.5	-6.1	-5.5	-7.2	-8.8	-9.0
15	-0.3	-0.1	-0.7	-3.1	-3.1	-3.9	-5.7	-5.5	-5.1	-4.0	-3.8	-3.5	-3.4	-5.5	-7.1	-7.1
16	0.2	1.7	1.4	0.4	-0.8	-2.6	-3.6	-3.7	-4.2	-3.5	-2.9	-3.6	-3.2	-4.5	-6.6	-5.3
17	-10.4	-11.0	-11.4	-12.7	-12.7	-14.0	-12.9	-9.5	-11.6	-8.5	-11.4	-8.5	-6.2	-4.5	-6.6	-8.4
18	-0.9	-0.4	-1.6	-1.1	-2.3	-3.8	-4.6	-3.6	-3.6	-2.6	-2.5	-2.6	-0.5	-1.2	-3.1	-3.6
19	0.6	0.7	-0.4	-1.0	-1.5	-3.7	-5.5	-5.4	-5.1	-4.1	-5.7	-4.5	-3.2	-4.5	-6.2	-4.0
20	-3.5	-2.7	-3.3	-3.7	-4.5	-5.9	-9.0	-9.0	-10.7	-11.7	-9.5	-8.6	-8.3	-8.4	-8.7	-10.2
21	-2.9	-2.2	-3.1	-2.5	-3.0	-5.1	-8.1	-8.2	-8.4	-8.5	-7.5	-6.2	-6.7	-7.9	-8.2	-8.2
22	-4.6	-3.5	-4.5	-4.7	-4.6	-7.0	-11.2	-12.2	-13.5	-11.7	-11.6	-10.7	-6.5	-7.7	-8.8	-10.2
23	-5.8	-4.5	-4.9	-4.2	-5.6	-8.3	-13.3	-14.2	-14.3	-15.6	-12.8	-13.3	-11.3	-9.8	-10.5	-11.6
24	-7.3	-5.5	-6.3	-5.8	-6.6	-9.8	-14.0	-15.9	-16.2	-16.5	-16.4	-14.4	-13.2	-13.7	-14.3	-14.3
25	-8.6	-7.2	-7.5	-6.5	-8.0	-11.1	-13.3	-17.9	-17.9	-17.9	-17.6	-17.2	-16.2	-15.2	-15.2	-16.1
26	-9.8	-9.1	-9.5	-10.0	-10.2	-13.7	-13.6	-20.9	-21.6	-20.4	-21.2	-20.4	-18.5	-17.5	-18.5	-18.5
27	-11.9	-11.8	-12.3	-12.6	-13.2	-16.8	-15.6	-24.4	-24.6	-23.2	-24.7	-22.7	-22.6	-22.1	-21.6	-22.8

TABLE III. - Concluded. SAMPLE OUTPUT FROM SUBROUTINE DADIFF

## C C M P A R I S C N O F T W O C A T A S E T S .

DATA SET ONE	DATA SET TWO
NATIONAL AERONAUTICS AND SPACE ADMINISTRATION	NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
LEWIS RESEARCH CENTER	LEWIS RESEARCH CENTER
PROPULSION SYSTEMS ACOUSTICS BRANCH	PROPULSION SYSTEMS ACOUSTICS BRANCH
SAMPLE NOISE DATA	SAMPLE NOISE DATA, SECOND SET
CONFIGURATION	CONFIGURATION
100	101
PERCENT SPEED	PERCENT SPEED
75.0	50.0
RPM	RPM
1800.	1200.
BANDS FROM	BANDS FROM
50 TO 20000 HERTZ	50 TO 20000 HERTZ
16 ANGLES EVERY 10. DEGREES FROM 10. TO 160.	16 ANGLES EVERY 10. DEGREES FROM 10. TO 160.

## PERCEIVED AND TONE-CORRECTED PERCEIVED NOISE LEVELS AND DIFFERENCES ALONG SIDELINES

59.0 F, 70.0 PERCENT RH

## 270. FT SIDELINE

ANGLE	10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.
SET TWO PNCE	62.6	74.7	79.5	83.0	82.7	80.8	79.2	80.2	82.7	83.7	84.0	85.1	85.0	80.9	74.4	65.3
SET ONE PNCE	69.3	80.1	85.3	89.3	85.2	88.9	88.6	88.9	92.7	93.0	93.3	93.2	92.4	88.6	83.1	74.8
DELTA PNCE	-6.7	-5.4	-5.8	-6.4	-6.6	-8.1	-9.4	-8.7	-10.0	-9.2	-9.3	-8.1	-7.5	-7.8	-8.7	-9.5
SET TWO PNLT	62.6	74.7	75.5	83.0	82.7	80.8	79.2	80.2	82.7	83.7	84.0	85.1	85.0	80.9	74.4	65.3
SET ONE PNLT	72.6	83.4	88.8	93.0	92.5	91.9	91.2	90.6	94.9	94.6	95.6	96.5	93.5	90.2	84.2	76.3
DELTA PNLT	-10.0	-8.7	-9.2	-10.1	-9.8	-11.1	-12.0	-10.3	-12.2	-10.9	-11.6	-9.7	-8.5	-9.3	-9.8	-11.0

## 1000. FT SIDELINE

ANGLE	10.	20.	30.	40.	50.	60.	70.	80.	90.	100.	110.	120.	130.	140.	150.	160.
SET TWO PNCE	41.0	58.3	64.4	68.7	69.1	67.9	66.7	68.1	70.1	71.4	71.4	72.4	71.7	66.5	55.8	48.3
SET ONE PNCE	49.5	65.0	71.7	76.4	76.9	76.9	76.6	76.2	79.7	79.5	80.3	75.8	76.6	74.3	69.2	60.3
DELTA PNCE	-8.5	-6.7	-7.3	-7.8	-7.8	-9.0	-9.9	-8.0	-9.6	-8.5	-8.9	-7.4	-6.8	-7.4	-9.4	-12.0
SET TWO PNLT	41.0	58.3	64.4	68.7	69.1	67.9	66.7	68.1	70.1	71.4	71.4	72.4	71.7	66.5	55.8	48.3
SET ONE PNLT	52.5	68.4	75.2	80.2	80.2	79.9	79.2	77.9	81.9	81.6	82.6	81.4	80.0	76.0	70.4	61.7
DELTA PNLT	-11.5	-10.1	-10.8	-11.5	-11.1	-12.1	-12.5	-9.7	-11.8	-10.3	-11.2	-9.1	-8.3	-9.1	-10.6	-13.4

TABLE IV. - LISTING OF TYPICAL SET OF WORKING DATA

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
 LEWIS RESEARCH CENTER  
 PULSIVE SYSTEMS ACOUSTICS BRANCH  
 SAMPLE NOISE DATA

100	180.0	75.0	27	16	3								
143.5	6.1	-18.9	-18.5	-15.2	-13.0	-13.3	-14.4	-15.4	-14.6	-15.1	-15.3		
		-15.7	-15.5	-14.7	-14.1	-13.6	-14.2	-5.7	-11.9	-10.0	-5.2		
		-8.1	-6.9	-5.5	-4.7	-4.6	-2.8	0.					
10.0	10.0	20.0	30.0	40.0	50.0	60.0	70.0	80.0	90.0	100.0	110.0		
120.0	130.0	140.0	150.0	160.0									
1	20	-2.1	-5.0	-6.2	-5.2	-2.5	-3.5	-3.9	-2.2	-4.7	-2.4		
		-1.5	-0.5	2.0	3.3	6.8	6.8						
2	63	-7.4	-6.1	-6.6	-7.5	-6.7	-6.7	-5.9	-5.6	-5.4	-4.9		
		-2.4	-0.3	2.4	4.4	7.3	8.3						
3	80	-9.5	-9.0	-10.8	-10.7	-10.0	-5.5	-9.7	-8.0	-6.3	-4.5		
		-2.0	-0.8	2.2	4.5	8.2	8.5						
4	100	-8.3	-7.8	-7.8	-6.4	-6.1	-8.6	-6.4	-5.4	-5.1	-2.3		
		-1.8	0.5	2.6	4.5	7.4	6.6						
5	125	-6.7	-7.2	-7.6	-7.1	-6.6	-5.1	-4.6	-3.4	-2.6	-1.4		
		-0.2	1.4	2.4	4.3	6.9	3.8						
6	160	-5.6	-4.0	-4.6	-4.3	-3.1	-2.8	-2.5	-2.3	-1.3	-0.8		
		-0.5	0.9	2.2	4.0	5.5	2.4						
7	200	-4.9	-4.0	-4.2	-4.4	-4.5	-5.0	-4.9	-4.5	-3.2	-2.4		
		-0.5	1.2	3.6	4.8	6.5	2.2						
8	250	-6.1	-5.6	-5.8	-5.6	-5.8	-5.3	-4.8	-3.6	-2.6	-1.1		
		0.4	2.0	3.2	4.6	5.6	2.5						
9	315	-5.9	-5.6	-4.3	-3.0	-4.4	-3.9	-3.8	-2.9	-1.6	-1.1		
		0.2	1.7	3.6	4.4	4.6	1.3						
10	400	-6.2	-5.2	-4.7	-3.7	-4.2	-2.9	-3.9	-3.4	-1.9	-0.9		
		0.4	2.2	3.6	4.4	4.1	-0.2						
11	500	-3.2	-2.1	-2.9	-2.4	-1.6	-3.1	-3.4	-2.4	-1.4	-0.4		
		0.4	1.7	3.3	3.6	3.5	-1.0						
12	630	-4.4	-3.3	-2.3	-0.8	-1.6	-2.5	-2.1	-1.6	-0.6	-0.1		
		0.9	1.6	3.0	2.7	1.7	-2.2						
13	800	-0.4	-1.2	-1.7	2.3	0.0	-3.0	-2.2	-0.8	-1.5	-0.2		
		0.2	0.9	2.5	1.8	0.8	-2.5						
14	1000	-0.8	-1.2	0.4	-0.3	2.7	-1.4	-1.4	-2.8	-2.1	-1.3		
		-0.1	1.0	2.2	2.1	0.6	-3.2						
15	1250	-0.5	1.0	1.4	2.5	0.8	-1.0	-1.1	-1.8	-2.0	-1.0		
		0.7	0.5	1.0	0.5	-0.3	-3.9						
16	1600	0.1	0.6	0.4	0.4	-0.4	-1.3	-2.1	-2.5	-1.5	-0.1		
		1.0	2.2	3.0	-0.0	-1.3	-4.7						
17	2000	2.4	3.5	3.9	4.9	2.2	0.3	-2.3	-5.0	-2.2	-3.8		
		0.7	-0.2	-0.8	-4.8	-6.7	-9.2						
18	2500	0.3	1.3	1.6	1.8	0.5	-1.2	-3.2	-2.7	-1.2	-0.7		
		1.3	2.4	1.8	-0.7	-2.7	-6.6						
19	3150	-2.2	-0.6	0.1	0.9	-1.1	-2.4	-3.6	-2.9	-0.7	0.1		
		1.6	3.6	3.1	-1.1	-3.4	-6.8						
20	4000	-3.2	-1.6	-1.0	0.0	-1.6	-4.3	-3.5	-4.3	1.4	1.5		
		1.7	2.3	3.0	0.9	-3.6	-6.3						
21	5000	-2.1	-0.6	-0.6	0.4	-0.6	-3.6	-3.9	-3.1	-1.1	-0.1		
		0.7	2.7	4.1	1.7	-1.0	-6.1						
22	6300	-3.3	-1.6	0.1	2.0	-2.1	-3.7	-3.9	-3.7	1.1	0.5		
		1.5	2.0	2.6	-0.2	-3.1	-7.7						
23	8000	-4.1	-2.5	-1.5	0.3	-2.5	-4.1	-4.0	-3.1	-0.3	2.4		
		1.9	2.9	3.2	0.0	-3.3	-8.1						
24	10000	-3.4	-2.0	-1.2	0.0	-2.9	-4.2	-4.2	-4.7	-0.9	0.8		
		1.8	3.9	3.5	1.0	-1.6	-7.8						
25	12500	-2.2	-1.4	-0.7	1.1	-1.4	-4.2	-4.0	-3.5	-1.2	0.3		
		1.3	3.2	3.0	1.3	-1.4	-7.6						
26	16000	-3.2	-1.5	0.0	1.3	-1.9	-4.4	-4.4	-4.2	-0.5	0.8		
		1.3	3.4	3.5	0.8	-2.2	-7.5						
27	20000	-2.3	-1.0	0.0	0.1	-1.4	-4.0	-2.4	-4.4	-1.4	0.4		
		1.0	3.2	3.5	1.1	-1.7	-7.0						

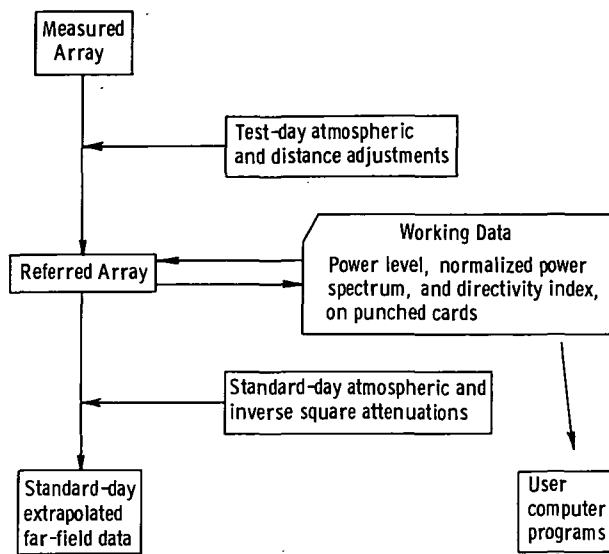


Figure 1. - Major elements of data handling system.

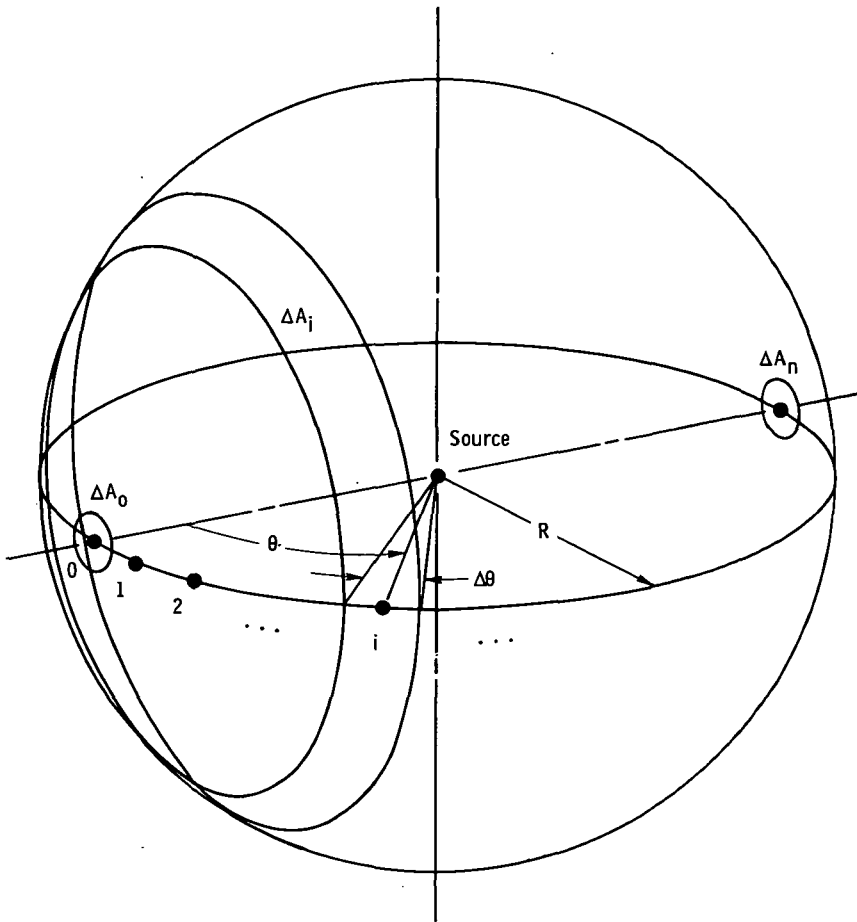


Figure 2 - Geometry of enclosing sphere for far-field acoustic measurements, showing elemental areas without ground plane.



POSTMASTER: If Undeliverable (Section 158  
Postal Manual) Do Not Return

*"The aeronautical and space activities of the United States shall be conducted so as to contribute . . . to the expansion of human knowledge of phenomena in the atmosphere and space. The Administration shall provide for the widest practicable and appropriate dissemination of information concerning its activities and the results thereof."*

—NATIONAL AERONAUTICS AND SPACE ACT OF 1958

## NASA SCIENTIFIC AND TECHNICAL PUBLICATIONS

**TECHNICAL REPORTS:** Scientific and technical information considered important, complete, and a lasting contribution to existing knowledge.

**TECHNICAL NOTES:** Information less broad in scope but nevertheless of importance as a contribution to existing knowledge.

**TECHNICAL MEMORANDUMS:** Information receiving limited distribution because of preliminary data, security classification, or other reasons. Also includes conference proceedings with either limited or unlimited distribution.

**CONTRACTOR REPORTS:** Scientific and technical information generated under a NASA contract or grant and considered an important contribution to existing knowledge.

**TECHNICAL TRANSLATIONS:** Information published in a foreign language considered to merit NASA distribution in English.

**SPECIAL PUBLICATIONS:** Information derived from or of value to NASA activities. Publications include final reports of major projects, monographs, data compilations, handbooks, sourcebooks, and special bibliographies.

**TECHNOLOGY UTILIZATION PUBLICATIONS:** Information on technology used by NASA that may be of particular interest in commercial and other non-aerospace applications. Publications include Tech Briefs, Technology Utilization Reports and Technology Surveys.

*Details on the availability of these publications may be obtained from:*

**SCIENTIFIC AND TECHNICAL INFORMATION OFFICE**

**NATIONAL AERONAUTICS AND SPACE ADMINISTRATION**

**Washington, D.C. 20546**