# 15. Motion of a Fragment in Disturbed Air Behind a Meteor Body 



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Even if we use the classical physical theory of meteors, there may still be some possibility for agreement between observations and theory. The effect of disturbed air behind the main body on the motion and ablation of fragments has not yet been considered. This effect may have importank consequences, observed partly as the differonce between the photometrically and the dynamically determined masses of the meteor body. By use of extreme mathematical conditions, this difference can be made to reach orders of magnitude during the latter part of the trajectory. However, the physical interpretation is considered only roughly in this paper, and the computed model needs further improvement. But the main purpose here-finding an important effect for the explanation of the discrepancy between the dynamic and the photometric masses, especially for large bodies-has been achieved.

## ASSUMPTIONS

The mathematical model used here contains the following simplifications: If a single meteor body with velocity $v$ and radius $R$ deposits its mass only in small fragments of radius $R_{f}$, the ablation of a spherical stony parent body is given by the following equations:

$$
\begin{align*}
\frac{d R}{d t} & =\frac{-\Lambda(t) \rho v^{3}}{8 Q \rho_{m}}  \tag{1}\\
\frac{d v}{d t} & =\frac{-3 \Gamma \rho v^{2}}{4 R \rho_{m}} \tag{2}
\end{align*}
$$

$$
\begin{equation*}
\frac{d \rho}{d t}=b \rho v \cos z_{R} \tag{3}
\end{equation*}
$$

and

$$
\begin{equation*}
\Lambda(t)=A_{0}+A_{1} \cdot t+A_{2} \cdot t^{2} \tag{4}
\end{equation*}
$$

where $Q=c \cdot \tau_{c}, \Lambda$ is the heat-transfer coefficient, $\rho$ is the density of the free atmosphere, $\rho_{m}$ is the density of the meteoroids, $c$ is the specific heat of the meteoroid, $b$ is the reciprocal of the density scale height of the atmosphere, and the temperaturd of the meteoroid outside the atmosphere is taken as zero. Equation (4) is simply an approximate polynomial expansion of second degree in time for the heat-transfer coefficient.

The following idealized history of each small fragment is assumed. The fragment leaves the parent body with temperature $\tau_{c}$. Starting with this temperature, the fragment is heated to the evaporation temperature, cooling by radiation being taken into account. Isothermal heating is assumed for such small fragments, and the following equations are thus valid for the first part of the trajectory:

$$
\begin{gather*}
\frac{c \cdot \rho_{m} R_{f}}{3} \frac{d T}{d t}+\sigma_{R}\left(T^{4}-T_{0}^{4}\right)=\frac{\Lambda_{f} \rho_{f} v^{3}}{8}  \tag{5}\\
\frac{d v}{d t}=\frac{-3 \Gamma \rho_{f} v^{2}}{4 R_{f} \rho_{m}}  \tag{6}\\
\frac{d H}{d t}=-v \cos z_{R} \tag{7}
\end{gather*}
$$

and

$$
\begin{equation*}
\rho_{f}=\rho-F\left(H_{0}-H\right) \tag{8}
\end{equation*}
$$

Here, $\sigma_{R}$ is the Stefan-Boltzmann constant, $\Lambda_{f}$ the heat-transfer coefficient for fragments, and $\Gamma$ the drag coefficient.

We deal with a very simplified case in this paper, the air density at height $H$ being the only disturbed entity; the density in the wake is denoted by $\rho_{f}$. In equation (8), $F$ describes this disturbance. The function $F$ also depends on the height of fragment separation, $H_{0}$; it has two extreme values: $F=0$ if the atmosphere behind the parent body is not disturbed, and $F=\rho$ if there is a vacuum behind.

After being heated to the evaporation temperature, the fragment starts the second part of its trajectory. The temperature is determined by radiation, vaporization, and aerodynamic heating. Mass loss and luminosity occur until the energy flux is balanced by radiation. The second part of the fragment's trajectory can be described by the following equation:

$$
\begin{equation*}
\frac{d R_{f}}{d t}=-\frac{\Lambda_{f} \rho_{f} v^{3}-4 \sigma_{R}\left(T^{4}-T_{0}^{4}\right)}{8 Q_{f} \rho_{m}} \tag{9}
\end{equation*}
$$

and by equations (6), (7), and (8). Here, $Q_{J}$ is the heat of ablation of vapors from the fragment. Visual radiated luminosity $I_{f}$ belonging to one fragment is then described in terms of the luminous efficiency coefficient $\tau_{0}$ by

$$
\begin{align*}
I_{f}= & 1 / 2 A\left(4 / 3 \pi \rho_{m}\right)^{2 / 3} \tau_{0} v^{3} R_{f}^{2}\left(H_{0}\right) \\
& \times\left[\sigma_{f} \Gamma \rho_{\rho} v^{3}-\frac{4 \sigma_{R}\left(T^{4}-T_{0}^{4}\right)}{Q_{f}}\right] \\
& \times \exp \left[-\frac{\sigma_{f}}{3}\left(v_{I I_{0}}{ }^{2}-v^{2}\right)\right] \tag{10}
\end{align*}
$$

where

$$
\sigma_{f} \equiv \frac{\Lambda_{f}}{2 \Gamma Q_{f}}
$$

Also, $A$ denotes the shape factor.

## PROCEDURE USED

The total light intensity at a given height $H_{m}$ is the sum of all the partial intensitics from all the fragments that separated from the parent body at heights $H \geq H_{m}$. It is clear that the fragments that scparated within heights $H_{1}>H \geq H_{m}$
are not shining, because they are not yet heated enough. The beginning of the light curve is taken at the height corresponding to the air density $\rho_{B}$ computed for the ending of the preheating period according to the formula by Ceplecha and Padevět (1961):

$$
\begin{equation*}
\rho_{B}=\frac{8 \lambda \tau(R)}{a v_{\infty}{ }^{3} R}\left(W R \frac{e^{W R}+e^{-W R}}{e^{W R}-e^{-W R}}-1\right) \tag{11}
\end{equation*}
$$

where $W=\left(b \cos z_{R} \cdot v_{\infty}\right)^{1 / 2} / \beta, \beta^{2}$ is the thermal diffusivity, $\lambda$ is the thermal conductivity of the meteoroid, $\tau$ is the temperature at which structural failure occurs, and $a$ is the accommodation cocfficient. The end of the light curve is defined to be at the height where the velocity has dccreased to $5 \mathrm{~km} / \mathrm{s}$.

The dynamic and photometric masses can now be computed and compared by means of the same procedures as for observed meteors. Equations (1) to (4) are used to find the radius and thus the mass of the body, and the conventional luminous equation

$$
\begin{equation*}
m_{p}(t)=\frac{2}{\tau_{0}} \int_{t}^{t_{\mathrm{ond}}} \frac{I\left(t^{\prime}\right)}{v^{3}} d t^{\prime}+m_{\mathrm{end}} \tag{12}
\end{equation*}
$$

is used to determine the photometric mass. Here, $v$ is the velocity of the parent body and $I$ is the instantaneous intensity. The integration constant $m_{\text {end }}$ was taken as the dynamically determined mass at the terminal point.

The change of air density behind the parent body might alter the computed photometric masses, but there is no need to attain higher values than those in the paper by McCrosky and Ceplecha (1970). However, if, for example, a constant factor for decreasing the density behind the main body is used, this undesired effect happens. But such a simple model for the function $F$ does not correspond even to a primitive physical guess. We would expect a decrease of the disturbance farther behind the meteor body, terminating with the density of the undisturbed atmosphere. It scems that each realistic model of the disturbance will result in a decrease of luminosity originating from one fragment. In an attempt to increase the photometric mass, we should find a model of the disturbance that would decrease the level of the photometric mass as little as possible and we should look for another


Figure 1.-Logarithm of the ratio of the photometric mass $m_{p}$ to the dynamical mass $m_{d}$ for an initial radius $R_{\infty}=100 \mathrm{~cm}$ and an initial velocity $v_{\infty}=12$ $\mathrm{km} / \mathrm{s}$.


Figure 3.-Logarithm of the ratio of the photometric mass $m_{p}$ to the dynamical mass $m_{d}$ for an initial rutius $R_{\infty}=100 \mathrm{~cm}$ and an initial velocity $v_{\infty}=30 \mathrm{~km} / \mathrm{s}$.


Figure 2.-Logarithm of the ratio of the photometric mass $m_{p}$ to the dynamical mass $m_{d}$ for an initial radius $R_{\infty}=1000 \mathrm{~cm}$ and an initial velocity $v_{\infty}=12 \mathrm{~km} / \mathrm{s}$.


Figure 4--Logarithm of the ratio of the photometric mass $m_{p}$ to the dynamical mass $m_{d}$ for an initial radius $R_{\infty}=1000 \mathrm{~cm}$ and an initial velocity $v_{\infty}=30 \mathrm{~km} / \mathrm{s}$.
effect strong enough to increase the level. The first condition may be met by a rather extreme and simple function-the discontinuous function $F=\rho$ for an interval of $H_{0}>H>H_{2}$, and $F=0$ for an interval of $H_{2} \geqq H \geqq H_{m}$. The fragments begin by moving in vacuum but cannot move ahead of the decelerating parent body. As an extreme case, we can imagine that the fragments are moving in a trap close behind the body, departing at height $H_{2}$. This extreme choice of the function $F$ serves the computations well as the first approximation of some more realistic model. The real function should start from a value very close to the air density (almost vacuum), followed by a steep change that is represented here by the discontinuous jump to zero. All our preliminary computations used this discontinuous function.

An additional parameter now is the number of fragments crossing the discontinuity in the air density at a given height $H_{m}$. If the discontinuous function $F$ is as above, only one change results: The height of the separation of the corresponding fragment will be higher than in the case without any disturbance. But we are interested in differences of orders of magnitude, and this is not possible with the above model with $\Lambda$ constant. On the other hand, if we assume that $\Lambda$ decreases with time, a steep decrease of the main-body mass results during the sccond part of the trajectory. Generally speaking, the decrease of $\Lambda$ with time can be expected from the physical point of view.

## RESULTS

Until now, only a limited number of cases have been computed. The numerical choice of parameters was made rather arbitrarily, with only great differences between the computed photometric and the computed dynamic masses being ac-
cepted. We can guess that the differences will be less after more physical factors are included, but the computed examples illustrate the general situation well.

Using the above model, we can explain the great difference between the dynamic and the photometric mass during the second part of the trajectory, if we assume chondritic densities for the body. As a function of time, the difference between the two masses is less in the first part of the trajectory and becomes greater at the end (see figs. 1 to 4).

## COMMENTS

$D=B_{0}+B_{1} \cdot H_{2}+B_{2} \cdot H_{2}{ }^{2}$ is the difference in height in kilometers between the fragment separation $\left(H_{0}\right)$ and that of departure from the trap $\left(H_{2}\right)$.
Numcrical parameters used for computation of the values in tables 1 to 4 (cgs) are as follows:

$$
\begin{array}{ll}
a & =1 \\
A & \\
b & \\
=1.21 \\
c & \\
=1.6 \times 10^{7} \mathrm{~cm}^{2} \mathrm{~s}^{-2} \mathrm{~cm}^{-1} \\
Q_{f} & =8 \times 10^{10} \mathrm{~cm}^{2} \mathrm{~s}^{-2} \\
\text { initial } R_{f} & =0.01 \mathrm{~cm} \\
T_{0} & =0^{\circ} \mathrm{K} \\
T_{B} & =2580^{\circ} \mathrm{K} \\
\cos z_{R} & =1 \\
\Gamma & =0.5 \\
\lambda & =3 \times 10^{5} \mathrm{~g} \mathrm{~cm} \mathrm{~s}^{-3} \mathrm{deg}^{-1} \\
\rho_{m} & =3.5 \mathrm{~g} \mathrm{~cm}^{-3} \\
& \\
\sigma_{f} & =\frac{\Lambda_{f}}{2 \Gamma Q_{f}}=2 \times 10^{-12} \mathrm{~cm}^{-2} \mathrm{~s}^{-2} \\
\tau_{c} & \\
& 600^{\circ} \mathrm{C}
\end{array}
$$

The results do not depend on the value $\tau_{0}$.

## REFERENCES

Ceplecha, Z., and Padevét, V., 1961. The beginning of rapid evaporation of meteors of different dimensions, Bull. Astron. Inst. Czech., 12, 191-195.
McCrosky, R. E., and Ceplecha, Z., 1970. Fireballs and the physical theory of meteors, Bull. Astron. Inst. Czech., 21, 271-296.

Table 1.-Meleoroid With Initial Velocity $\nabla_{\infty}=12 \mathrm{~km} / \mathrm{s}$ and Radius $\mathrm{R}_{\infty}=100 \mathrm{~cm}$

| $\begin{aligned} & \text { Velocity } v \\ & (\mathrm{~km} / \mathrm{s}) \end{aligned}$ | Dynamical mass $m_{d}$ (g) | $\underset{(\mathrm{km})}{\text { Height } H}$ | Epoch $t$ <br> (s) | Assumed heat-transfer coefficient A |
| :---: | :---: | :---: | :---: | :---: |
| 5.000 | $6.0967 \times 10^{2}$ | 17.654 | 4.2592 | 0.00224 |
| 5.692 | $6.4917 \times 10^{2}$ | 17.999 | 4.1959 | 0.01734 |
| 6.446 | $7.8346 \times 10^{2}$ | 18.363 | 4.1370 | 0.03138 |
| 7.225 | $1.0994 \times 10^{3}$ | 18.750 | 4.0816 | 0.04458 |
| 7.980 | $1.7900 \times 10^{3}$ | 19.161 | 4.0286 | 0.05720 |
| 8.669 | $3.2799 \times 10^{3}$ | 19.601 | 3.9768 | 0.06952 |
| 9.268 | $6.4496 \times 10^{3}$ | 20.072 | 3.9253 | 0.08178 |
| 9.772 | $1.3035 \times 10^{4}$ | 20.570 | 3.8728 | 0.09426 |
| 10.191 | $2.6252 \times 10^{4}$ | 21.111 | 3.8184 | 0.10719 |
| 10.536 | $5.1745 \times 10^{4}$ | 21.706 | 3.7610 | 0.12082 |
| 10.822 | $9.9022 \times 10^{4}$ | 22.365 | 3.6994 | 0.13545 |
| 11.058 | $1.8352 \times 10^{5}$ | 23.104 | 3.6322 | 0.15140 |
| 11.255 | $3.2981 \times 10^{5}$ | 23.945 | 3.5576 | 0.16911 |
| 11.421 | $5.7708 \times 10^{5}$ | 24.920 | 3.4727 | 0.18922 |
| 11.561 | $9.8699 \times 10^{5}$ | 26.079 | 3.3737 | 0.21267 |
| 11.680 | $1.6587 \times 10^{6}$ | 27.506 | 3.2540 | 0.24100 |
| 11.780 | $2.7585 \times 10^{6}$ | 29.358 | 3.1011 | 0.27714 |
| 11.866 | $4.5839 \times 10^{6}$ | 31.994 | 2.8875 | 0.32754 |
| 11.939 | $7.7595 \times 10^{6}$ | 36.419 | 2.5258 | 0.41270 |
| 12.000 | $1.4661 \times 10^{7}$ | 73.536 | 0.0000 | 1.00000 |

Table 2.-Meteoroid With Initial Velocity $\mathrm{v}_{\infty}=12 \mathrm{~km} / \mathrm{s}$ and Radius $\mathrm{R}_{\infty}=1000 \mathrm{~cm}$

| $\begin{gathered} \text { Velocity } v \\ (\mathrm{~km} / \mathrm{s}) \end{gathered}$ | Dynamical mass $m_{d}$ (g) | $\begin{aligned} & \text { Height } H \\ & (\mathrm{~km}) \end{aligned}$ | Epoch $t$ <br> (s) | Assumed heat-transfer coefficient $\Lambda$ |
| :---: | :---: | :---: | :---: | :---: |
| 5.000 | $4.3224 \times 10^{4}$ | 1.7160 | 5.2882 | 0.05170 |
| 6.281 | $8.8988 \times 10^{4}$ | 2.2542 | 5.2282 | 0.06718 |
| 7.454 | $2.4031 \times 10^{5}$ | 2.8161 | 5.1762 | 0.08050 |
| 8.408 | $7.1522 \times 10^{5}$ | 3.4026 | 5.1285 | 0.09263 |
| 9.145 | $2.0631 \times 10^{6}$ | 4.0220 | 5.0826 | 0.10425 |
| 9.708 | $5.4479 \times 10^{6}$ | 4.6686 | 5.0370 | 0.11574 |
| 10.144 | $1.3075 \times 10^{7}$ | 5.3541 | 4.9903 | 0.12742 |
| 10.487 | $2.8694 \times 10^{7}$ | 6.0834 | 4.9419 | 0.13949 |
| 10.762 | $5.8364 \times 10^{7}$ | 6.8585 | 4.8907 | 0.15215 |
| 10.987 | $1.1141 \times 10^{8}$ | 7.6891 | 4.8360 | 0.16561 |
| 11.174 | $2.0175 \times 10^{8}$ | 8.5869 | 4.7766 | 0.18011 |
| 11.330 | $3.4984 \times 10^{8}$ | 9.5653 | 4.7112 | 0.19596 |
| 11.463 | $5.8537 \times 10^{8}$ | 10.643 | 4.6380 | 0.21355 |
| 11.576 | $9.5153 \times 10^{8}$ | 11.699 | 4.5544 | 0.23343 |
| 11.674 | $1.5114 \times 10^{9}$ | 12.860 | 4.4564 | 0.25647 |
| 11.759 | $2.3604 \times 10^{9}$ | 14.281 | 4.3374 | 0.28406 |
| 11.832 | $3.6457 \times 10^{9}$ | 16.114 | 4.1851 | 0.31878 |
| 11.896 | $5.6144 \times 10^{3}$ | 18.698 | 3.9716 | 0.36624 |
| 11.952 | $8.7454 \times 10^{9}$ | 23.053 | 3.6087 | 0.44380 |
| 12.000 | $1.4661 \times 10^{10}$ | 73.532 | 0.0000 | 1.00000 |

Table 3.-Meteoroid With Initial Velocity $\mathrm{v}_{\infty}=30 \mathrm{~km} / \mathrm{s}$ and Radius $\mathrm{R}_{\infty}=100 \mathrm{~cm}$

| $\begin{aligned} & \text { Velocity } v \\ & (\mathrm{~km} / \mathrm{s}) \end{aligned}$ | Dynamical mass $m_{d}$ <br> (g) | $\underset{(\mathrm{km})}{\text { Height } H}$ | Epoch $t$ (s) | Assumed heat-transfer coefficient A |
| :---: | :---: | :---: | :---: | :---: |
| 5.000 | $3.2649 \times 10^{-5}$ | 29.104 | 1.7436 | 0.01609 |
| 23.680 | $7.7116 \times 10^{-1}$ | 29.455 | 1.7215 | 0.02887 |
| 26.363 | $2.8707 \times 10^{1}$ | 29.826 | 1.7074 | 0.03705 |
| 27.493 | $2.2584 \times 10^{2}$ | 30.220 | 1.6934 | 0.04514 |
| 28.144 | $9.9065 \times 10^{2}$ | 30.641 | 1.6789 | 0.05352 |
| 28.571 | $3.1699 \times 10^{3}$ | 31.091 | 1.6637 | 0.06230 |
| 28.874 | $8.3382 \times 10^{3}$ | 31.576 | 1.6476 | 0.07159 |
| 29.101 | $1.9233 \times 10^{4}$ | 32.096 | 1.6303 | 0.08154 |
| 29.276 | $4.0281 \times 10^{4}$ | 32.646 | 1.6117 | 0.09226 |
| 29.416 | $7.8608 \times 10^{4}$ | 33.250 | 1.5914 | 0.10396 |
| 29.530 | $1.4512 \times 10^{5}$ | 33.919 | 1.5691 | 0.11683 |
| 29.623 | $2.5621 \times 10^{5}$ | 34.675 | 1.5442 | 0.13114 |
| 29.701 | $4.3758 \times 10^{5}$ | 35.540 | 1.5161 | 0.14733 |
| 29.766 | $7.2682 \times 10^{5}$ | 36.548 | 1.4837 | 0.16591 |
| 29.822 | $1.1836 \times 10^{6}$ | 37.757 | 1.4455 | 0.18784 |
| 29.870 | $1.9020 \times 10^{6}$ | 39.260 | 1.3988 | 0.21462 |
| 29.910 | $3.0356 \times 10^{6}$ | 41.235 | 1.3387 | 0.24899 |
| 29.945 | $4.8632 \times 10^{8}$ | 44.094 | 1.2542 | 0.24899 0.29726 |
| 29.975 | $7.9709 \times 10^{6}$ | 49.320 | 1.1101 | $0.37931$ |
| 30.000 | $1.4661 \times 10^{7}$ | 87.079 | 0.0000 | $\begin{aligned} & 0.37931 \\ & 1.00000 \end{aligned}$ |

Table 4.-Meteoroid With Inilial Velocity $\mathrm{v}_{\infty}=30 \mathrm{~km} / \mathrm{s}$ and Radius $\mathrm{R}_{\infty}=1000 \mathrm{~cm}$

| Velicity $v$ <br> $(\mathrm{~km} / \mathrm{s})$ | Dynamical mass $m_{d}$ <br> $(\mathrm{~g})$ | Height $H$ <br> $(\mathrm{~km})$ | Epoch $t$ <br> $(\mathrm{~s})$ | Assumed <br> heat-transfer <br> coefficient $A$ |
| :---: | :---: | :---: | :---: | :---: |
|  |  |  |  |  |
| 5.000 | $8.6440 \times 10^{-3}$ | 13.510 |  |  |
| 24.240 | $2.6803 \times 10^{3}$ | 13.855 | 2.2721 | 0.01375 |
| 26.458 | $6.9058 \times 10^{4}$ | 14.29 | 2.2333 | 0.03189 |
| 27.473 | $4.5678 \times 10^{8}$ | 14.605 | 2.2193 | 0.03842 |
| 28.083 | $1.7743 \times 10^{6}$ | 15.016 | 2.2052 | 0.04498 |
| 28.496 | $5.1904 \times 10^{8}$ | 15.455 | 2.1907 | 0.05173 |
| 28.799 | $1.2789 \times 10^{7}$ | 15.928 | 2.1755 | 0.05881 |
| 29.030 | $2.7879 \times 10^{7}$ | 16.438 | 2.1593 | 0.06634 |
| 29.211 | $5.5661 \times 10^{7}$ | 16.993 | 2.1420 | 0.07438 |
| 29.358 | $1.0420 \times 10^{8}$ | 17.600 | 2.1234 | 0.08303 |
| 29.479 | $1.8520 \times 10^{8}$ | 18.272 | 2.1030 | 0.09246 |
| 29.579 | $3.1676 \times 10^{8}$ | 19.024 | 2.0807 | 0.10280 |
| 29.664 | $5.2456 \times 10^{8}$ | 19.875 | 2.0557 | 0.11433 |
| 29.736 | $8.4820 \times 10^{8}$ | 20.838 | 2.0276 | 0.12732 |
| 29.798 | $1.3466 \times 10^{9}$ | 2.976 | 1.9951 | 0.14225 |
| 29.851 | $2.1108 \times 10^{9}$ | 23.377 | 1.9568 | 0.15984 |
| 29.897 | $3.2897 \times 10^{9}$ | 25.198 | 1.9101 | 0.18126 |
| 29.937 | $5.1470 \times 10^{9}$ | 27.790 | 1.8499 | 0.20871 |
| 29.971 | $8.2266 \times 10^{9}$ | 32.286 | 1.7652 | 0.24718 |
| 30.000 | $1.4661 \times 10^{10}$ | 87.077 | 1.6207 | 0.31234 |

