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# 15. Motion of a Fragment in Disturbed Air Behind a Meteor Body

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 $E^{\rm ven \ if \ we \ use}$  the classical physical theory of meteors, there may still be some possibility for agreement between observations and theory. The effect of disturbed air behind the main body on the motion and ablation of fragments has not yet been considered. This effect may have important consequences, observed partly as the difference between the photometrically and the dynamically determined masses of the meteor body. By use of extreme mathematical conditions, this difference can be made to reach orders of magnitude during the latter part of the trajectory. However, the physical interpretation is considered only roughly in this paper, and the computed model needs further improvement. But the main purpose here-finding an important effect for the explanation of the discrepancy between the dynamic and the photometric masses, especially for large bodies—has been achieved.

#### ASSUMPTIONS

The mathematical model used here contains the following simplifications: If a single meteor body with velocity v and radius R deposits its mass only in small fragments of radius  $R_f$ , the ablation of a spherical stony parent body is given by the following equations:

$$\frac{dR}{dt} = \frac{-\Lambda(t)\rho v^3}{8Q\rho_m} \tag{1}$$

$$\frac{dv}{dt} = \frac{-3\Gamma\rho v^2}{4R\rho_m} \tag{2}$$

$$\frac{d\rho}{dt} = b\rho v \cos z_R \tag{3}$$

and

$$\Lambda(t) = A_0 + A_1 \cdot t + A_2 \cdot t^2 \tag{4}$$

where  $Q = c \cdot \tau_c$ ,  $\Lambda$  is the heat-transfer coefficient,  $\rho$  is the density of the free atmosphere,  $\rho_m$  is the density of the meteoroids, c is the specific heat of the meteoroid, b is the reciprocal of the density scale height of the atmosphere, and the temperature of the meteoroid outside the atmosphere is taken as zero. Equation (4) is simply an approximate polynomial expansion of second degree in time for the heat-transfer coefficient.

The following idealized history of each small fragment is assumed. The fragment leaves the parent body with temperature  $\tau_e$ . Starting with this temperature, the fragment is heated to the evaporation temperature, cooling by radiation being taken into account. Isothermal heating is assumed for such small fragments, and the following equations are thus valid for the first part of the trajectory:

$$\frac{\rho_m R_f}{3} \frac{dT}{dt} + \sigma_R (T^4 - T_0^4) = \frac{\Lambda_f \rho_f v^3}{8} \qquad (5)$$

$$\frac{dv}{dt} = \frac{-3\Gamma\rho_f v^2}{4R_f \rho_m} \tag{6}$$

$$\frac{dH}{dt} = -v\cos z_R \tag{7}$$

(8)

 $\rho_f = \rho - F(H_0 - H)$ 

and

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Here,  $\sigma_R$  is the Stefan-Boltzmann constant,  $\Lambda_f$  the heat-transfer coefficient for fragments, and  $\Gamma$  the drag coefficient.

We deal with a very simplified case in this paper, the air density at height H being the only disturbed entity; the density in the wake is denoted by  $\rho_f$ . In equation (8), F describes this disturbance. The function F also depends on the height of fragment separation,  $H_0$ ; it has two extreme values: F = 0 if the atmosphere behind the parent body is not disturbed, and  $F = \rho$  if there is a vacuum behind.

After being heated to the evaporation temperature, the fragment starts the second part of its trajectory. The temperature is determined by radiation, vaporization, and aerodynamic heating. Mass loss and luminosity occur until the energy flux is balanced by radiation. The second part of the fragment's trajectory can be described by the following equation:

$$\frac{dR_f}{dt} = -\frac{\Lambda_f \rho_f v^3 - 4\sigma_R (T^4 - T_0^4)}{8Q_f \rho_m}$$
(9)

and by equations (6), (7), and (8). Here,  $Q_f$  is the heat of ablation of vapors from the fragment. Visual radiated luminosity  $I_f$  belonging to one fragment is then described in terms of the luminous efficiency coefficient  $\tau_0$  by

$$I_{f} = \frac{1}{2}A \left(\frac{4}{3}\pi\rho_{m}\right)^{2/3} \tau_{0} v^{3} R_{f}^{2}(H_{0})$$

$$\times \left[\sigma_{f} \Gamma \rho_{f} v^{3} - \frac{4\sigma_{R}(T^{4} - T_{0}^{4})}{Q_{f}}\right]$$

$$\times \exp\left[-\frac{\sigma_{f}}{3} \left(v_{H_{0}}^{2} - v^{2}\right)\right]$$
(10)

where

$$\sigma_f \equiv \frac{\Lambda_f}{2\Gamma Q_f}$$

Also, A denotes the shape factor.

#### **PROCEDURE USED**

The total light intensity at a given height  $H_m$ is the sum of all the partial intensities from all the fragments that separated from the parent body at heights  $H \ge H_m$ . It is clear that the fragments that separated within heights  $H_1 > H \ge H_m$  are not shining, because they are not yet heated enough. The beginning of the light curve is taken at the height corresponding to the air density  $\rho_B$ computed for the ending of the preheating period according to the formula by Ceplecha and Padevět (1961):

$$\rho_B = \frac{8\lambda\tau(R)}{av_{\infty}^3 R} \left( WR \frac{e^{WR} + e^{-WR}}{e^{WR} - e^{-WR}} - 1 \right) \quad (11)$$

where  $W = (b \cos z_R \cdot v_{\infty})^{1/2} / \beta$ ,  $\beta^2$  is the thermal diffusivity,  $\lambda$  is the thermal conductivity of the meteoroid,  $\tau$  is the temperature at which structural failure occurs, and a is the accommodation coefficient. The end of the light curve is defined to be at the height where the velocity has decreased to 5 km/s.

The dynamic and photometric masses can now be computed and compared by means of the same procedures as for observed meteors. Equations (1) to (4) are used to find the radius and thus the mass of the body, and the conventional luminous equation

$$m_{p}(t) = \frac{2}{\tau_{0}} \int_{t}^{t_{\text{end}}} \frac{I(t')}{v^{3}} dt' + m_{\text{end}} \qquad (12)$$

is used to determine the photometric mass. Here, v is the velocity of the parent body and I is the instantaneous intensity. The integration constant  $m_{end}$  was taken as the dynamically determined mass at the terminal point.

The change of air density behind the parent body might alter the computed photometric masses, but there is no need to attain higher values than those in the paper by McCrosky and Ceplecha (1970). However, if, for example, a constant factor for decreasing the density behind the main body is used, this undesired effect happens. But such a simple model for the function F does not correspond even to a primitive physical guess. We would expect a decrease of the disturbance farther behind the meteor body, terminating with the density of the undisturbed atmosphere. It seems that each realistic model of the disturbance will result in a decrease of luminosity originating from one fragment. In an attempt to increase the photometric mass, we should find a model of the disturbance that would decrease the level of the photometric mass as little as possible and we should look for another



FIGURE 1.—Logarithm of the ratio of the photometric mass  $m_p$  to the dynamical mass  $m_d$  for an initial radius  $R_{\infty} = 100$  cm and an initial velocity  $v_{\infty} = 12$  km/s.



FIGURE 3.—Logarithm of the ratio of the photometric mass  $m_p$  to the dynamical mass  $m_d$  for an initial radius  $R_{\infty} = 100$  cm and an initial velocity  $v_{\infty} = 30$  km/s.



FIGURE 2.—Logarithm of the ratio of the photometric mass  $m_p$  to the dynamical mass  $m_d$  for an initial radius  $R_{\infty} = 1000$  cm and an initial velocity  $v_{\infty} = 12$  km/s.



FIGURE 4.—Logarithm of the ratio of the photometric mass  $m_p$  to the dynamical mass  $m_d$  for an initial radius  $R_{\infty} = 1000$  cm and an initial velocity  $v_{\infty} = 30$  km/s.

effect strong enough to increase the level. The first condition may be met by a rather extreme and simple function-the discontinuous function  $F = \rho$  for an interval of  $H_0 > H > H_2$ , and F = 0 for an interval of  $H_2 \ge H \ge H_m$ . The fragments begin by moving in vacuum but cannot move ahead of the decelerating parent body. As an extreme case, we can imagine that the fragments are moving in a trap close behind the body, departing at height  $H_2$ . This extreme choice of the function F serves the computations well as the first approximation of some more realistic model. The real function should start from a value very close to the air density (almost vacuum), followed by a steep change that is represented here by the discontinuous jump to zero. All our preliminary computations used this discontinuous function.

An additional parameter now is the number of fragments crossing the discontinuity in the air density at a given height  $H_m$ . If the discontinuous function F is as above, only one change results: The height of the separation of the corresponding fragment will be higher than in the case without any disturbance. But we are interested in differences of orders of magnitude, and this is not possible with the above model with  $\Lambda$  constant. On the other hand, if we assume that  $\Lambda$  decreases with time, a steep decrease of the main-body mass results during the second part of the trajectory. Generally speaking, the decrease of  $\Lambda$  with time can be expected from the physical point of view.

### RESULTS

Until now, only a limited number of cases have been computed. The numerical choice of parameters was made rather arbitrarily, with only great differences between the computed photometric and the computed dynamic masses being accepted. We can guess that the differences will be less after more physical factors are included, but the computed examples illustrate the general situation well.

Using the above model, we can explain the great difference between the dynamic and the photometric mass during the second part of the trajectory, if we assume chondritic densities for the body. As a function of time, the difference between the two masses is less in the first part of the trajectory and becomes greater at the end (see figs. 1 to 4).

## **COMMENTS**

 $D=B_0+B_1\cdot H_2+B_2\cdot H_2^2$  is the difference in height in kilometers between the fragment separation  $(H_0)$  and that of departure from the trap  $(H_2)$ .

Numerical parameters used for computation of the values in tables 1 to 4 (cgs) are as follows:

a	=1
A	= 1.21
b	$= 1.6 \times 10^{-6} \text{ cm}^{-1}$
с	$= 10^7 \text{ cm}^2 \text{ s}^{-2} \text{ deg}^{-1}$
$Q_f$	$=8 \times 10^{10} \text{ cm}^2 \text{ s}^{-2}$
initial i	$R_f = 0.01 \text{ cm}$
$T_0$	$=0^{\circ}$ K
$T_B$	=2580° K
$\cos z_R$	= 1
Г	=0.5
λ	$=3 \times 10^{5} \text{ g cm s}^{-3} \text{ deg}^{-1}$
$\rho_m$	$=3.5 \text{ g cm}^{-3}$
σſ	$= \frac{\Lambda_f}{2\Gamma Q_f} = 2 \times 10^{-12} \mathrm{cm}^{-2} \mathrm{s}^{-2}$
τ.	= 600° C

The results do not depend on the value  $\tau_0$ .

#### REFERENCES

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Velocity v (km/s)	Dynamical mass $m_d$ (g)	Height H (km)	Epoch t (s)	Assumed hcat-transfer coefficient Λ
5.000	$ \begin{array}{c} 6.0967 \times 10^2 \\ 6.4917 \times 10^2 \end{array} $	$17.654 \\ 17.999$	$\begin{array}{c} 4.2592\\ 4.1959\end{array}$	0.00 <b>22</b> 4 0.01734
6.446	7.8346×10 <sup>2</sup>	18.363	4.1370	0.03138
7.225	$1.0994 \times 10^{3}$	18.750	4.0816	0.04458
7.980	$1.7900 \times 10^{3}$	19.101	3.9768	0.06952
8.009 9.268	6.4496×10 <sup>3</sup>	20.072	3.9253	0.08178
9.772	$1.3035  imes 10^{4}$	20.570	3.8728	0.09426
10.191	$2.6252  imes 10^4$	21.111	3.8184	0.10719
10.536	$5.1745 \times 10^{4}$	21.700	3.6994	0.13545
10.822	$9.9022 \times 10^{5}$ 1.8352 × 10 <sup>5</sup>	23.104	3.6322	0.15140
11.255	3.2981×10 <sup>5</sup>	23.945	3.5576	0.16911
11.421	5.7708×105	24.920	3.4727	0.18922
11.561	9.8699×10 <sup>5</sup>	26.079	3.3737	0.21267
11.680	$1.6587 \times 10^{6}$	27.500	3 1011	0.27714
11.780	$2.7585 \times 10^{\circ}$ $4.5830 \times 10^{\circ}$	25.558	2.8875	0.32754
11.800	7.7595×10 <sup>6</sup>	36.419	2.5258	0.41270
12.000	1.4661×107	73.536	0.0000	1.00000

TABLE 1.—Meleoroid With Initial Velocity  $v_{\infty} = 12$  km/s and Radius  $R_{\infty} = 100$  cm

TABLE 2.—Meteoroid With Initial Velocity  $v_{\infty} = 12$  km/s and Radius  $R_{\infty} = 1000$  cm

Velocity v (km/s)	Dynamical mass m <sub>d</sub> (g)	Height <i>H</i> (km)	Epoch t (s)	Assumed heat-transfe coefficient A
5.000	4.3224×104	1.7160	5.2882	0.05170
6.281	8.8988×104	2.2542	5.2282 5.4700	0.00113
7.454	$2.4031 \times 10^{5}$	2.8161	5.1702	0.00000
8.408	7.1522×105	3.4026	5.1285	0.09205
9.145	$2.0631 \times 10^{6}$	4.0220	5.0826	0.10425
9.708	5.4479×10 <sup>6</sup>	4.6686	5.0370	0.11574
10.144	1.3075×107	5.3541	4.9903	0.12742
10.487	2.8694×107	6.0834	4.9419	0.15949
10.762	5.8364×107	6.8585	4,8907	0.15215
10.987	1.1141×10 <sup>8</sup>	7.6891	4.8360	0.10001
11.174	$2.0175 \times 10^{8}$	8.5869	4.7766	0.18011
11.330	3.4984×10 <sup>8</sup>	9.5653	4.7112	0,19590
11.463	5.8537×10 <sup>8</sup>	10.643	4.6380	0.21355
11.576	9.5153×10 <sup>8</sup>	11.699	4.5544	0.23343
11.674	$1.5114 \times 10^{9}$	12.860	4.4564	0.25647
11.759	2.3604×10 <sup>9</sup>	14.281	4.3374	0.28406
11.832	$3.6457 \times 10^{9}$	16.114	4.1851	0.31878
11.896	5.6144×10°	18.698	3.9716	0.36624
11.952	8.7454×10°	23.053	3.6087	0.44380
12.000	1.4661×10 <sup>10</sup>	73.532	0.0000	1.00000

Velocity v (km/s)	Dynamical mass $m_d$ (g)	Height H (km)	Epoch t (s)	Assumed heat-transfer coefficient $\Lambda$
$\begin{array}{c} 5.000\\ 23.680\\ 26.363\\ 27.493\\ 28.144\\ 28.571\\ 28.874\\ 29.101\\ 29.276\\ 29.416\\ 29.530\\ 29.623\\ 29.701\\ 29.766\\ 29.822\\ 29.870\\ 29.910\\ 29.945\\ 29.975\\ 30.000\\ \end{array}$	$\begin{array}{c} 3.2649 \times 10^{-5} \\ 7.7116 \times 10^{-1} \\ 2.8707 \times 10^{1} \\ 2.2584 \times 10^{2} \\ 9.9065 \times 10^{2} \\ 3.1699 \times 10^{3} \\ 8.3382 \times 10^{3} \\ 1.9233 \times 10^{4} \\ 4.0281 \times 10^{4} \\ 7.8608 \times 10^{4} \\ 1.4512 \times 10^{5} \\ 2.5621 \times 10^{5} \\ 4.3758 \times 10^{5} \\ 7.2682 \times 10^{5} \\ 1.1836 \times 10^{9} \\ 1.9020 \times 10^{9} \\ 3.0356 \times 10^{6} \\ 4.8632 \times 10^{5} \\ 1.9709 \times 10^{5} \\ 1.4661 \times 10^{7} \end{array}$	$\begin{array}{c} 29.104\\ 29.455\\ 29.826\\ 30.220\\ 30.641\\ 31.091\\ 31.576\\ 32.096\\ 32.646\\ 33.250\\ 33.919\\ 34.675\\ 35.540\\ 36.548\\ 37.757\\ 39.260\\ 41.235\\ 44.094\\ 49.320\\ 57.270\\ 37.070\\ 57.070\\$	$\begin{array}{c} 1.7436\\ 1.7215\\ 1.7074\\ 1.6934\\ 1.6637\\ 1.6637\\ 1.6637\\ 1.66476\\ 1.6303\\ 1.6117\\ 1.5914\\ 1.5691\\ 1.5442\\ 1.5161\\ 1.4837\\ 1.4455\\ 1.3988\\ 1.3387\\ 1.2542\\ 1.1101\\ \end{array}$	$\begin{array}{c} 0.01609\\ 0.02887\\ 0.03705\\ 0.04514\\ 0.05352\\ 0.06230\\ 0.07159\\ 0.08154\\ 0.09226\\ 0.10396\\ 0.11683\\ 0.13114\\ 0.14733\\ 0.16591\\ 0.18784\\ 0.21462\\ 0.24899\\ 0.29726\\ 0.37931 \end{array}$
	1.1001 \ 10'	81.019	0.000	1.00000

TABLE 3.—Meteoroid With Initial Velocity  $v_{\infty}$  = 30 km/s and Radius  $R_{\infty}$  = 100 cm

TABLE 4.—Meteoroid With Initial Velocity v  $_{\infty}\!=\!30$  km/s and Radius R  $_{\infty}\!=\!1000$  cm

Velicity v (km/s)	Dynamical mass m <sub>d</sub> (g)	Height <i>H</i> (km)	Epoch t (s)	Assumed heat-transfer coefficient A
5.000	8.6440×10-3	13.510	2.2721	0.01375
24.240	$2.6803 \times 10^{3}$	13.855	2.2333	0.03189
26.458	$6.9058  imes 10^4$	14.219	2.2193	0.03842
27.473	4.5678×10 <sup>5</sup>	14.605	2.2052	0.04498
28.083	$1.7743 \times 10^{6}$	15.016	2.1907	0.05173
28.496	$5.1904  imes 10^{6}$	15.455	2.1755	0.05881
28.799	$1.2789 \times 10^{7}$	15.928	2.1593	0.06634
29.030	2.7879×107	16.438	2.1420	0.07438
29.211	5.5661×107	16.993	2.1234	0.08303
29.358	1.0420×10 <sup>8</sup>	17.600	2.1030	0.09246
29.479	$1.8520  imes 10^8$	18.272	2.0807	0.10280
29.579	$3.1676  imes 10^8$	19.024	2.0557	0.11433
29.664	$5.2456  imes 10^8$	19.875	2.0276	0.12732
29.736	$8.4820 \times 10^{8}$	20.838	1,9951	0.14225
29.798	1.3466×10°	21.976	1.9568	0,15984
29.851	2.1108×10°	23.377	1.9101	0.18126
29.897	3.2897×10 <sup>3</sup>	25.198	1.8499	0.20871
29.937	5.1470×10 <sup>9</sup>	27.790	1.7652	0.24718
29.971	8.2266×10 <sup>9</sup>	32.286	1.6207	0.31234
30.000	1.4661×1010	87.077	0.0000	1 00000