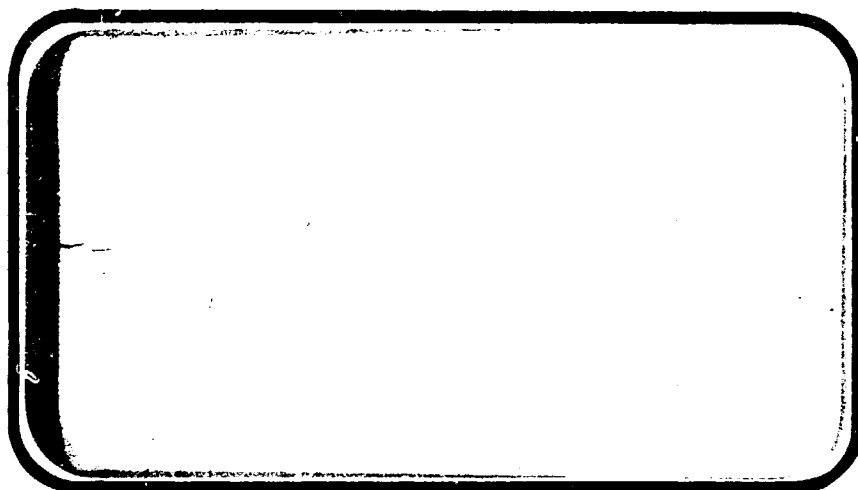


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NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

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NASA-CR-134080) SUPERSONIC PERFORMANCE,
STABILITY AND CONTROL CHARACTERISTICS OF
0.01875 SCALE MODEL ROCKWELL
INTERNATIONAL 089E-139B CREIFF
CORP.) 74 P HC \$6.75

Chrysler
CI 22B

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63/31 Unclas
32694

SPACE SHUTTLE

AEROTHERMODYNAMIC DATA REPORT

JOHNSON SPACE CENTER

HOUSTON, TEXAS

DATA Management services

SPACE DIVISION



CHRYSLER
CORPORATION

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SUPERSONIC PERFORMANCE, STABILITY AND
CONTROL CHARACTERISTICS OF A 0.01875 SCALE MODEL
ROCKWELL INTERNATIONAL 089B-139B
ORBITER CONFIGURATION
(LA8C)

By

R. W. Powell, NASA/LaRC
G. M. Ware, NASA/LaRC

Prepared under NASA Contract Number NAS9-13247

by

Data Management Services
Chrysler Corporation Space Division
New Orleans, La. 70189

for

Engineering Analysis Division

Johnson Space Center
National Aeronautics and Space Administration
Houston, Texas

WIND TUNNEL TEST SPECIFICS

Test Numbers: UPWT 1040
NASA Series No.: La8C
Date: July 10 - 13, (42 Occ. Hrs.)

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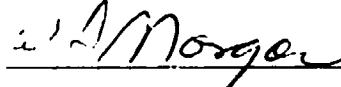
Phone: (804) 827-3984


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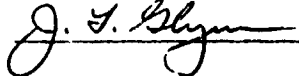
B. W. Myers
Data Operations





This document has been reviewed and is approved for release.

 F. N. D. Kemp
Data Management Services



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SUPERSONIC PERFORMANCE, STABILITY AND CONTROL CHARACTERISTICS
OF A 0.01875 SCALE MODEL ROCKWELL INTERNATIONAL
089B-139B ORBITER CONFIGURATION

By

R. W. Powell and G. M. Ware, NASA/LaRC

SUMMARY

An investigation was made in the Langley Unitary Plan Wind Tunnel at Mach numbers of 1.9 and 2.86 to study the supersonic aerodynamic characteristics of a Rockwell International shuttle orbiter configuration. Tests were made at a Reynolds number of 1.5×10^6 per foot with an angle-of-attack range of -4° to 28° and sideslip variations of -6° to 8° . The effects of elevon and aileron deflections were investigated.

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COEFFICIENT SCHEDULE:

A: CA, CN, CL, CLM, L/D, CD VS. ALPHA
 CN, CL VS. CLM
 CD VS. CL

C: CY, CYN, CBL VS. BETA

D: DCY/DA, DCYNDA, DCBLDA VS. ALPHA

E: DCY/DR, DCYNDR, DCBLDR VS. ALPHA

NOMENCLATURE
General

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
a		speed of sound; m/sec, ft/sec
C _p	CP	pressure coefficient; $(p_1 - p_\infty)/q$
M	MACH	Mach number; V/a
p		pressure; N/m^2 , psf
q	Q(NSM) Q(PSF)	dynamic pressure; $1/2\rho V^2$, N/m^2 , psf
RN/L	RN/L	unit Reynolds number; per m, per ft
V		velocity; m/sec, ft/sec
α	ALPHA	angle of attack, degrees
β	BETA	angle of sideslip, degrees
ψ	PSI	angle of yaw, degrees
ϕ	PHI	angle of roll, degrees
ρ		mass density; kg/m^3 , slugs/ft ³

Reference & C.G. Definitions

A _b		base area; m ² , ft ²
b	BREF	wing span or reference span; m, ft
c.g.		center of gravity
$\frac{l}{c}$ _{REF}	LREF	reference length or wing mean aerodynamic chord; m, ft
S	SREF	wing area or reference area; m ² , ft ²
	MRP	moment reference point
	XMRP	moment reference point on X axis
	YMRP	moment reference point on Y axis
	ZMRP	moment reference point on Z axis

SUBSCRIPTS

b	base
l	local
s	static conditions
t	total conditions
∞	free stream

NOMENCLATURE (Continued)

Body-Axis System

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
C_N	CN	normal-force coefficient; $\frac{\text{normal force}}{qS}$
C_A	CA	axial-force coefficient; $\frac{\text{axial force}}{qS}$
C_Y	CY	side-force coefficient; $\frac{\text{side force}}{qS}$
C_{A_b}	CAB	base-force coefficient; $\frac{\text{base force}}{qS}$ $-A_b(p_b - p_\infty)/qS$
C_{A_f}	CAF	forebody axial force coefficient, $C_A - C_{A_b}$
C_m	CLM	pitching-moment coefficient; $\frac{\text{pitching moment}}{qS l_{REF}}$
C_n	CYN	yawing-moment coefficient; $\frac{\text{yawing moment}}{qS b}$
C_l	CBL	rolling-moment coefficient; $\frac{\text{rolling moment}}{qS b}$

Stability-Axis System

C_L	CL	lift coefficient; $\frac{\text{lift}}{qS}$
C_D	CD	drag coefficient; $\frac{\text{drag}}{qS}$
C_{D_b}	CDB	base-drag coefficient; $\frac{\text{base drag}}{qS}$
C_{D_f}	CDF	forebody drag coefficient; $C_D - C_{D_b}$
C_Y	CY	side-force coefficient; $\frac{\text{side force}}{qS}$
C_m	CL	pitching-moment coefficient; $\frac{\text{pitching moment}}{qS l_{REF}}$
C_n	CLN	yawing-moment coefficient; $\frac{\text{yawing moment}}{qS b}$
C_l	CSL	rolling-moment coefficient; $\frac{\text{rolling moment}}{qS b}$
L/D	L/D	lift-to-drag ratio; C_L/C_D

NOMENCLATURE (Continued)

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
δ_{eL}		left elevon surface deflection angle, positive deflection trailing edge down; degrees.
δ_{eR}		right elevon surface deflection angle, positive deflection trailing edge down; degrees.
δ_e	ELEVTR	elevator, surface deflection angle, positive deflection trailing edge down, degrees, $(\delta_{eL} + \delta_{eR})/2$
δ_a	AILRON	aileron, aileron deflection angle, degrees, $(\delta_{eL} - \delta_{eR})/2$
	GT-LOC	grit location (refer to Test Conditions).
K	K	roughness height.
K/l	K/L	ratio of roughness to model body length ($l = 24.93$ in).
	CPC	cavity pressure coefficient.
	CPB1, CPB2	base pressure coefficients.
$C_{Y\beta}$	DCY/DB	side force coefficient derivative with respect to beat. Algebraic difference of the side force coefficient of two runs divided by the algebraic difference of the side slip angle of the runs; per degree.
$C_{n\beta}$	DCYNDB	yawing moment coefficient derivative with respect to beta. Algebraic difference of the yawing moment coefficient of two runs divided by the algebraic difference of the side slip angle of the runs; body axis system; per degree.
δ_R	RUDDER	rudder deflection angle, degree.
δ_{BF}	BDFLAP	body flap deflection angle, degree.
δ_{RF}	RUDFLF	rudder flare, split rudder deflection angle, degree.

NOMENCLATURE (Concluded)

<u>SYMBOL</u>	<u>SADSAC SYMBOL</u>	<u>DEFINITION</u>
$C_{l\beta}$	DCBLDB	rolling moment coefficient derivative with respect to beta. Algebraic difference of rolling moment coefficient of two runs divided by algebraic difference of side slip angle of the runs; body axis system; per degree.
$C_{y\delta_a}$	DCY/DA	side force coefficient derivative with respect to total aileron deflection. Algebraic difference of the side force coefficients of two runs divided by the algebraic difference of the total aileron deflection angle of the runs; per degree.
$C_{l\delta_a}$	DCBLDA	rolling moment coefficient derivative with respect to total aileron deflection. Algebraic difference of the rolling moment coefficient of two runs divided by the algebraic difference of the total aileron deflection angle of the runs; body axis system; per degree.
$C_{n\delta_a}$	DCYNDA	yawing moment coefficient derivative with respect to total aileron deflection. Algebraic difference of the yawing moment coefficient of two runs divided by the algebraic difference of the total aileron deflection angle of the runs; body axis system; per degree.
$C_{y\delta_r}$	DCY/DR	side force coefficient derivative with respect to rudder deflection. Algebraic difference of the side force coefficient of two runs divided by the algebraic difference of the rudder deflection angle of the runs; body axis system; per degree.
$C_{n\delta_r}$	DCYNDR	yawing moment coefficient derivative with respect to rudder deflection. Algebraic difference of the yawing moment coefficient of two runs divided by the algebraic difference of the rudder deflection angle of the runs; body axis system; per degree.
$C_{l\delta_r}$	DCBLDR	rolling moment coefficient derivative with respect to rudder deflection. Algebraic difference of the rolling moment coefficient of two runs divided by the algebraic difference of the rudder deflection angle of the runs; body axis system; per degree.

TEST FACILITY DESCRIPTION

The NASA LaRC 4-foot Unitary Plan Wind Tunnel (UPWT) is a closed-circuit, continuous flow, variable density facility. The test section is 4 feet by 4 feet long.

Two tunnel legs are available for supersonic testing in the Mach number ranges 1.47 to 2.86 (leg No. 1) and 2.29 to 4.63 (Leg No. 2). Leg No. 1 was used for this test. An asymmetric, sliding block nozzle position and total pressure setting provide the test Mach numbers at a specified Reynolds number. Reynolds number can be varied from 0.76 to 7.78 million per foot. Available stagnation pressure variation is 4.0 to 142. psia. Dynamic pressure variation is 95. to 1260. psf with normal operating stagnation temperature about 150°F in Mach modes 2 or 3 and about 175°F in Mach mode 4. The tunnel is equipped with a dry air supply, an evacuating system, and a cooling system. The facility power is approximately 83,000 horsepower.

Model mounting provisions consist of various sting arrangements, including axial (longitudinal), lateral (independent pitch and yaw), and roll movement with side wall support. A Schlieren system and oil flow visualization equipment are available. Data are recorded at the tunnel and reduced off-line at the Langley Computer Center. The tunnel is used for force and moment, pressure, and dynamic stability tests. Hot and cold jet effects and heat transfer have been studied in the UPWT.

CONFIGURATION INVESTIGATED

The configuration tested was a 0.01875 scale model of a blend of Rockwell International shuttle configurations. The model consisted of a 089B configuration with a 139B configuration nose forward of F.S. 500. A sketch of the model is shown in figure 2. All of the tests were made with the rudder flared to form a 40° wedge vertical tail and the body flap deflected -14.25°. Tests were made with elevon deflections ranging from -30° to 0°, and a 10° aileron deflection about a -10° elevon deflection.

DATA REDUCTION

A LaRC 832-B six-component strain gage balance was used to measure model forces and moments. All final data were presented along a set of body and stability axes passing through the nominal center of gravity located at F.S. 1076.48 or 65 percent of the body length. Model data were converted to standard NASA Coefficients using the following constants:

Reference Area, S_{ref} = wing planform area = 0.9457 ft.²

Reference Length, \bar{c}_{ref} = wing mean aerodynamic chord = 8.9025 in.

Reference Span, b_{ref} = wing span = 17.5628 in.

Transition was fixed with number 50 grit located 0.283 inch streamwise on wing and vertical tail, and 1.2 inches streamwise on nose. The drag data presented herein is gross drag in that base drag is included. Tabulated base pressure coefficients are presented, however, if corrections are desired.

TABLE II.

TEST: UPIUT 1040 (UARS)		DATA SET/RUN NUMBER COLLATION SUMMARY										DATE: July 10-13, 1973			
DATA SET IDENTIFIER	CONFIGURATION	SCHD.		GAIT		PARAMETERS/VALUES						NO. OF RUNS	MACH NUMBERS		
		α	β	SIZE	LOC	EL	RE	RF	SR	OR	NO.		NO.		
201	OSB w/ mod NBE + DMS	A	0	529	2	0	0	M25	40	0		1	1.6	1.9	2.86
202		T	3	T	T	0	0	T	T	T		2			
203		0	0	T	T	70	70	T	T	T		3			
204		3	0	T	T	70	70	T	T	T		4			
205		0	0	T	T	-20	-20	T	T	T		5			
206		F	0	T	T	-20	-20	T	T	T		6			
207		T	0	T	T	-20	-20	T	T	T		7			
208		F	0	T	T	-20	-20	T	T	T		8			
209		0	0	T	T	-30	-30	T	T	T		9			
210		F	0	T	T	-30	-30	T	T	T		10			
211		T	0	T	T	-30	-30	T	T	T		11			
212		A	3	T	T	0	-20	T	T	T		12			
213		15	C	T	T	0	-20	T	T	T		13			
214		20	T	T	T	0	-20	T	T	T		14			
215		25	T	T	T	0	-20	T	T	T		15			
216		15	T	T	T	-10	-10	T	T	T		16			
217		20	T	T	T	-10	-10	T	T	T		17			
218		25	C	T	T	-10	-10	M25	40	0		18			

1 7 13 19 25 31 37 43 49 55 61 67 75 76

CA...BA...KBL...KYN...GX...CPL...GPR...GCG... IDVAR (1) IDVAR (2) NDV

α OR β SCHEDULES A - 1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18

COEFFICIENTS C) - 6 7 8 9 10 11 12 13 14 15 16 17 18

TABLE III.
MODEL COMPONENT DIMENSIONAL DATA

MODEL COMPONENT: BODY - 089B-139B(Modified Nose)

GENERAL DESCRIPTION: Nose section from full-scale station 238.0 to STA. 500
from NAR drawing VL70-000139B. Remaining body AFT of STA 500 from NAR
drawing VL70-000093

DRAWING NUMBER: VL70-000093

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
		.01875
Length	1290.3	24.193
Max. Width	265.0	4.969
Max. Depth	248.0	4.650
Fineness Ratio	4.869	4.869
Area		
Max. Cross-Sectional	456.40	0.1605
Planform	_____	_____
Wetted	_____	_____
Base	_____	_____

TABLE III. (CONTINUED)

MODEL COMPONENT: ELEVON

GENERAL DESCRIPTION: CONFIGURATION PER LINES VL70-000093

DATA FOR (1) OF (2) SIDES

MODEL SCALE = 0.01875

DRAWING NUMBER: VL70-000093

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area	<u>205.517</u>	<u>0.0723</u>
Span (equivalent)	<u>353.34</u>	<u>6.625</u>
Inb'd equivalent chord	<u>114.78</u>	<u>2.152</u>
Outb'd equivalent chord	<u>55.00</u>	<u>1.031</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u>.208</u>	<u>.208</u>
At Outb'd equiv. chord	<u>.400</u>	<u>.400</u>
Sweep Back Angles, degrees		
Leading Edge	<u>0.00</u>	<u>0.00</u>
Tailing Edge	<u>-10.02</u>	<u>-10.02</u>
Hingeline	<u>0.00</u>	<u>0.00</u>
Area Moment (Normal to hinge line)- Ft ³	<u>1548.07</u>	<u>0.0102</u>

TABLE III. (CONTINUED)

MODEL COMPONENT: WING

GENERAL DESCRIPTION: Orbiter Configuration per Lines VL70-000093.

NOTE: (Dihedral angle is defined at the lower surface of the wing at the 75.33%
element line projected into a plane perpendicular to the FRL).

SCALE MODEL = 0.01875

DRAWING NUMBER: VL70-000093

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
<u>TOTAL DATA</u>		
Area		
Planform	2690.00	0.9457
Wetted	-----	-----
Span (equivalent)	936.68	17.56
Aspect Ratio	2.265	2.265
Rate of Taper	1.177	1.177
Taper Ratio	0.200	0.200
Dihedral Angle, degrees	3.500	3.500
Incidence Angle, degrees	3.000	3.000
Aerodynamic Twist, degrees	+3.000	+3.000
Toe-In Angle	-----	-----
Cant Angle	-----	-----
Sweep Back Angles, degrees	-----	-----
Leading Edge	45.000	45.000
Trailing Edge	-10.24	-10.24
0.25 Element Line	35.209	35.209
Chords:		
Root (Wing Sta. 0.0)	689.24	12.923
Tip, (equivalent)	137.85	2.585
MAC	474.81	8.903
Fus. Sta. of .25 MAC	1136.89	21.317
W.P. of .25 MAC	299.20	5.610
B.L. of .25 MAC	182.13	3.415
Airfoil Section	-----	-----
Root	-----	-----
Tip	-----	-----
<u>EXPOSED DATA</u>		
Area	1752.29	0.6160
Span, (equivalent)	720.68	13.513
Aspect Ratio	2.058	2.058
Taper Ratio	0.2451	0.2451
Chords		
Root	562.40	10.545
Tip	137.85	2.585
MAC	393.03	7.369
Fus. Sta. of .25 MAC	1185.31	22.224
W.P. of .25 MAC	300.20	5.629
B.L. of .25 MAC	143.76	2.700

TABLE III. (CONTINUED)

MODEL COMPONENT: Vertical Tail

GENERAL DESCRIPTION: Centerline vertical tail double wedge airfoil with rounded leading edge.

Scale Model = 0.01875

DRAWING NUMBER: VL70-000095

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area (theo) ft. ²	<u>413.25</u>	<u>0.145</u>
Span (equivalent)	<u>315.72</u>	<u>5.920</u>
Inb'd equivalent chord	<u>268.50</u>	<u>5.034</u>
Outb'd equivalent chord	<u>108.47</u>	<u>2.034</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u> </u>	<u> </u>
At Outb'd equiv. chord	<u> </u>	<u> </u>
Sweep Back Angles, degrees		
Leading Edge	<u>45</u>	<u>45</u>
Tailing Edge	<u>26.249</u>	<u>26.249</u>
Hingeline	<u> </u>	<u> </u>
Area Moment (Normal to hinge line)	<u> </u>	<u> </u>

TABLE III. (CONCLUDED)

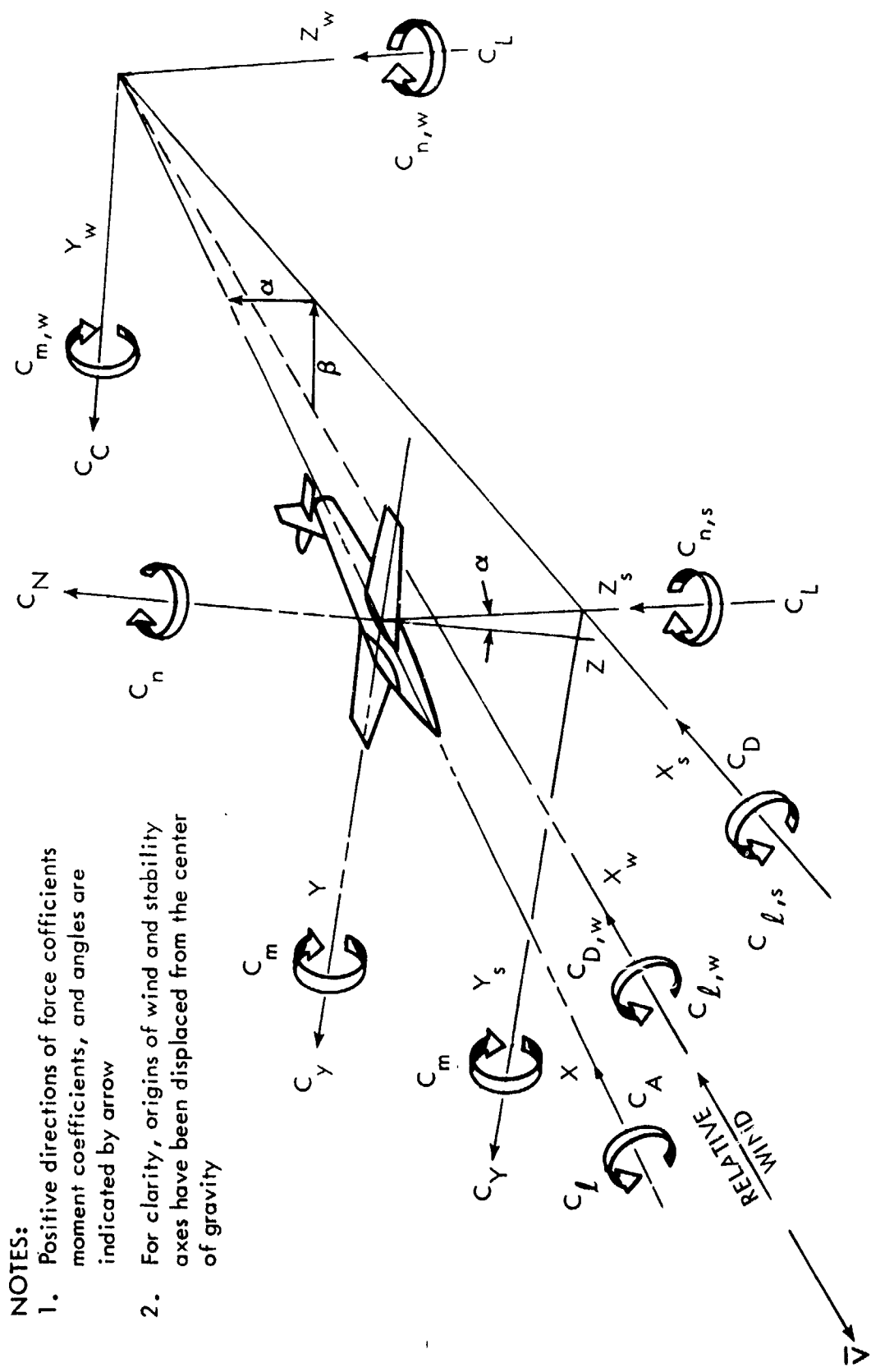
MODEL COMPONENT: RUDDER

GENERAL DESCRIPTION: CONFIGURATION PER LINES VL70-000095

SCALE MODEL = 0.01875

DRAWING NUMBER: VL70-000095

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area	<u>106.38</u>	<u>0.0374</u>
Span (equivalent)	<u>201.0</u>	<u>3.769</u>
Inb'd equivalent chord	<u>91.585</u>	<u>1.717</u>
Outb'd equivalent chord	<u>50.833</u>	<u>0.953</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u>0.400</u>	<u>0.400</u>
At Outb'd equiv. chord	<u>0.400</u>	<u>0.400</u>
Sweep Back Angles, degrees		
Leading Edge	<u>34.83</u>	<u>34.83</u>
Tailing Edge	<u>26.25</u>	<u>26.25</u>
Hingeline	<u>34.83</u>	<u>34.83</u>
Area Moment (Normal to hinge line)-Ft ³	<u>526.125</u>	<u>0.0034</u>



- NOTES:**
1. Positive directions of force coefficients moment coefficients, and angles are indicated by arrow
 2. For clarity, origins of wind and stability axes have been displaced from the center of gravity

Figure 1. - Axis Systems.

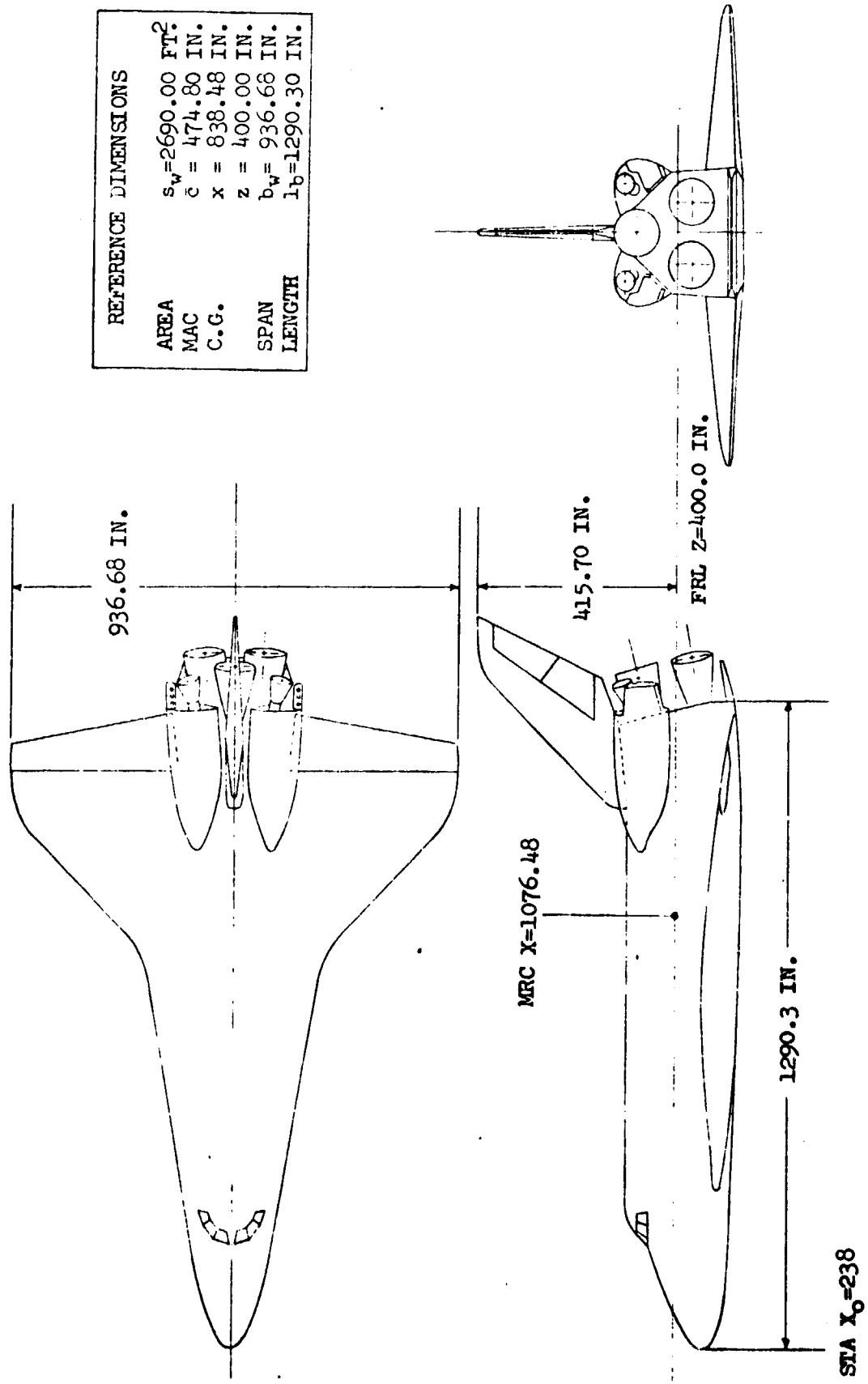
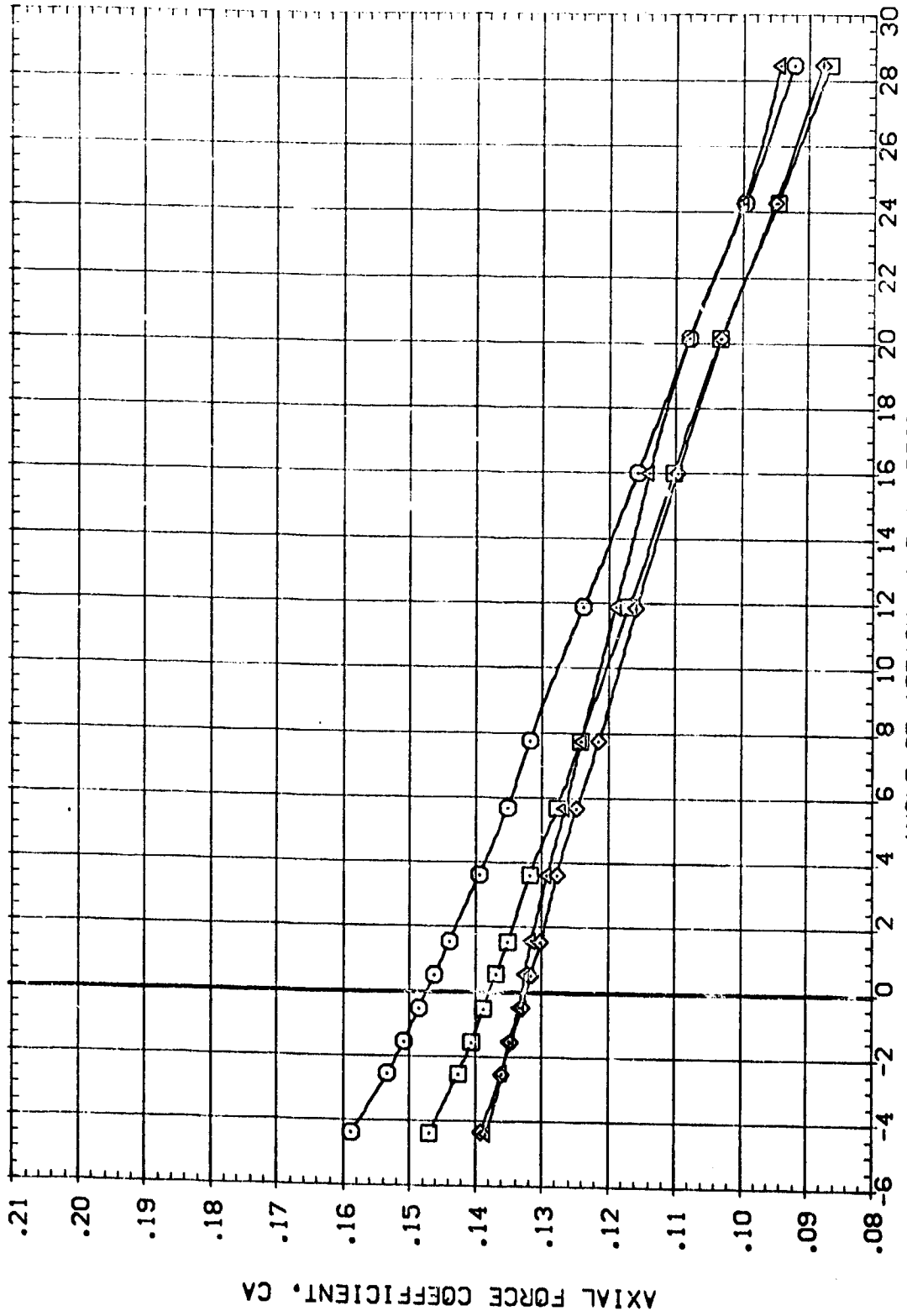


Figure 2. - SSV Orbiter Configuration.

DATA FIGURES

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	ELEVTR	AILTRON	BDFLAP	REFERENCE INFORMATION
(D-6208)	LA-8C: UPWT1040; CRBITER 0898 V/MOD: NOSE +CMS	3.000	-30.000	.000	-14.250	SRE 136.1808 SQ. IN.
(D-6205)	LA-8C: UPWT1040; CRBITER 0898 V/MOD: NOSE +CMS	3.000	-20.000	.000	-14.250	LREF 8.6025 INCHES
(D-6204)	LA-8C: UPWT1040; CRBITER 0898 V/MOD: NOSE +CMS	3.000	-10.000	.000	-14.250	BREF 17.5628 INCHES
(D-6202)	LA-8C: UPWT1040; CRBITER 0898 V/MOD: NOSE +CMS	3.000	.000	.000	-14.250	YMRP 15.7313 INCHES
						ZMRP .0000 INCHES
						SCALE .0189 INCHES

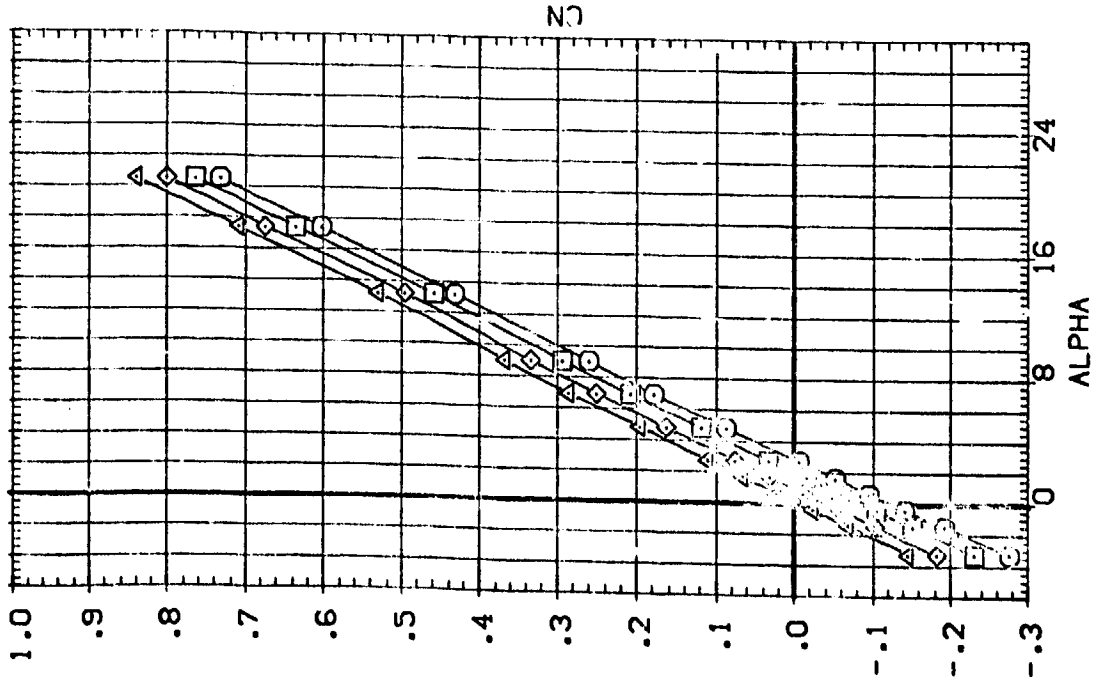
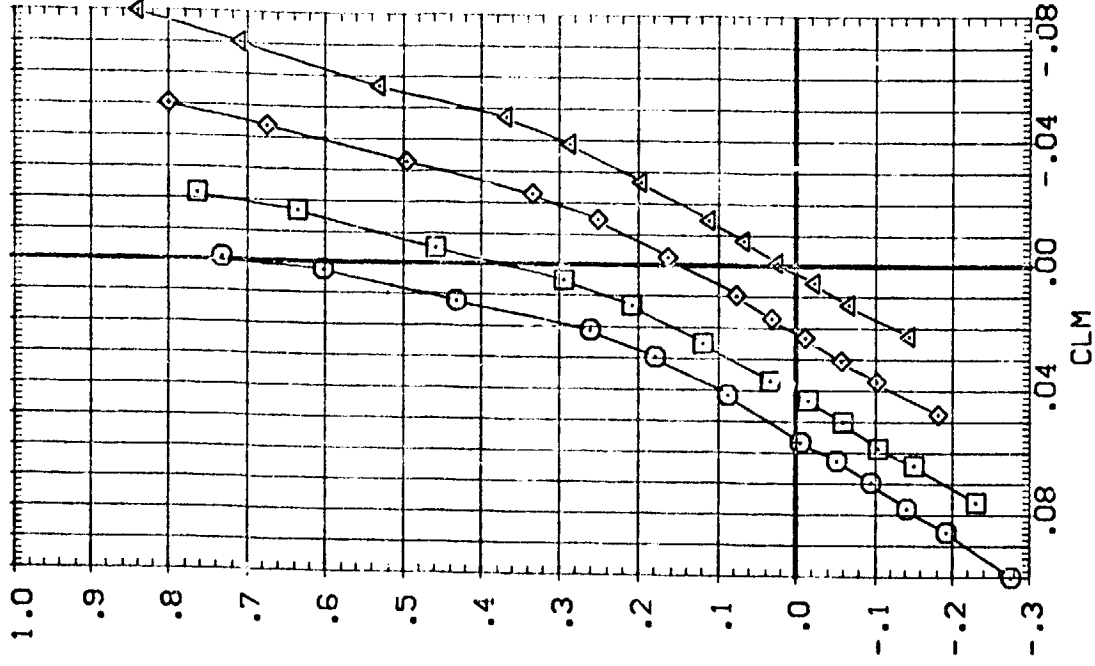


EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(B)MACH = 2.86

DATA SET SYMBOL: (DP6208) (DP6206) (DP6204) (DP6202)

CONFIGURATION DESCRIPTION	BETA	ELEVTR	AILRON	BOFLAP	REFERENCE INFORMATION
LA-8C: UPV11040: CRBITER 0899 V/MCD: NOSE +0MS	3.000	-30.000	.000	-14.250	SREF 136.1808 SQ. IN
LA-8C: UPV11040: CRBITER 0899 V/MCD: NOSE +0MS	3.000	-20.000	.000	-14.250	REF 8.9025 INCHES
LA-8C: UPV11040: CRBITER 0899 V/MCD: NOSE +0MS	3.000	-10.000	.000	-14.250	BRF 17.5628 INCHES
LA-8C: UPV11040: CRBITER 0899 V/MCD: NOSE +0MS	3.000	-10.000	.000	-14.250	YMRP 15.7313 INCHES
					ZMRP .0000 INCHES
					SCALE .0188 INCHES

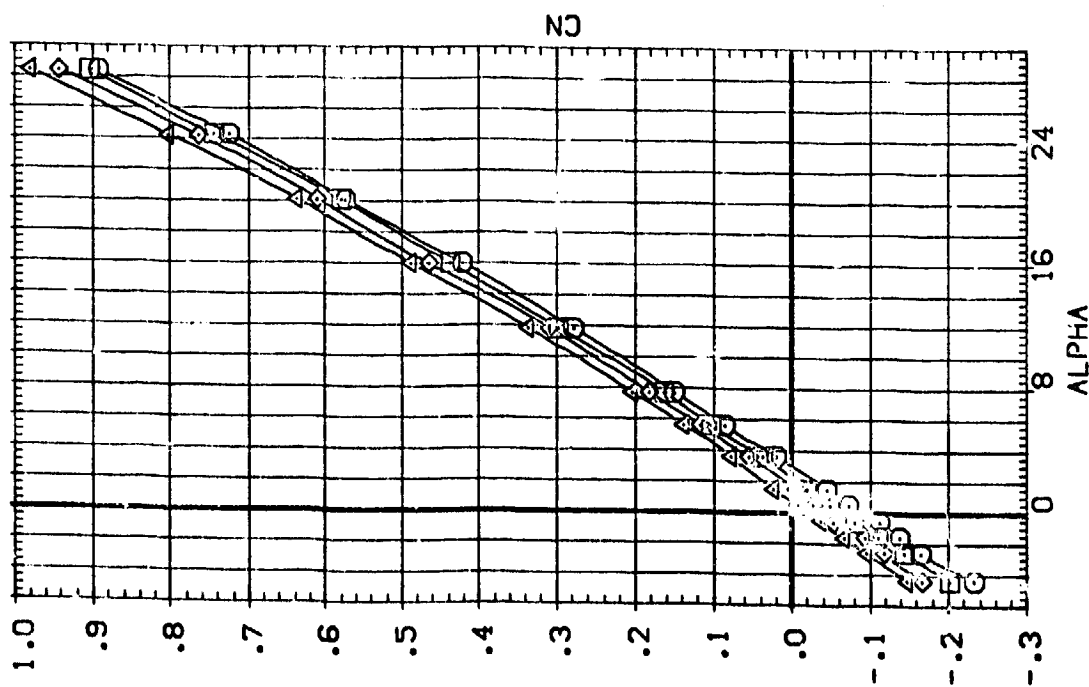
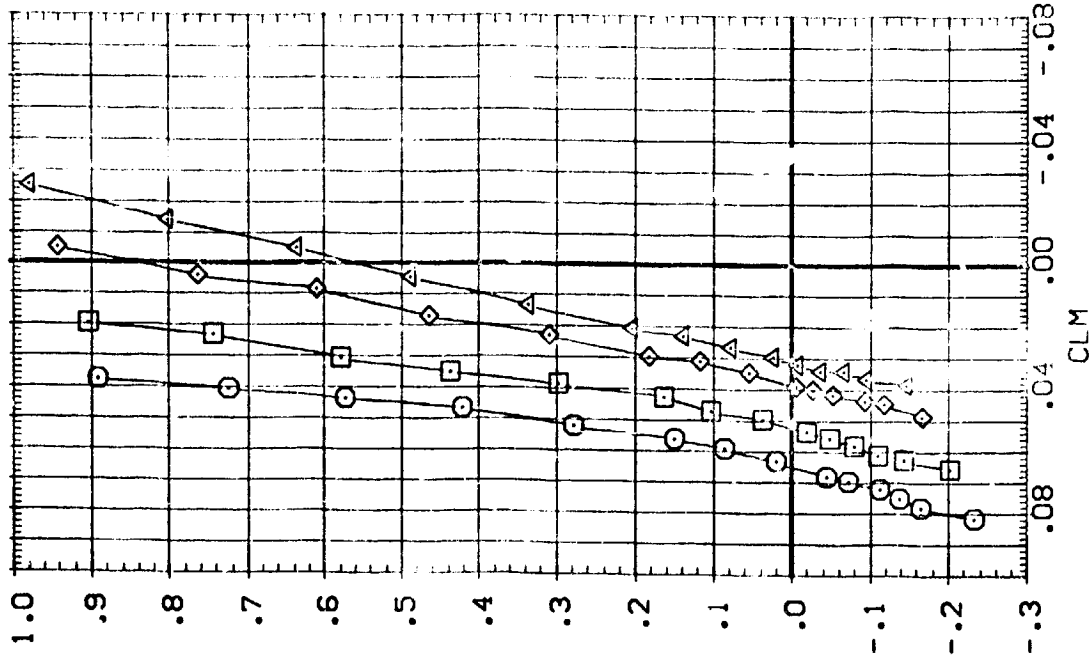


EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(A)MACH = 1.90

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	W/ MOD.	NOSE	+ONS
(DP6208)	LA-8C: UPVT1040. CRB11TER D898	V/ MOD.	NOSE	+ONS
(DP6206)	LA-8C: UPVT1040. CRB11TER D898	V/ MOD.	NOSE	+ONS
(DP6204)	LA-8C: UPVT1040. CRB11TER D898	V/ MOD.	NOSE	+ONS
(DP6202)	LA-8C: UPVT1040. CRB11TER D898	V/ MOD.	NOSE	+ONS

BETA	ELEVTR	AILRON	BOFLAP	REFERENCE INFORMATION
3.000	-30.000	.000	-14.250	SREF 136.1808 SQ. IN.
3.000	-20.000	.000	-14.250	LREF 8.9226 INCHES
3.000	-10.000	.000	-14.250	BREF 17.5828 INCHES
3.000	.000	.000	-14.250	XMRP 15.7313 INCHES
				YMRP .0000 INCHES
				ZMRP .0000 INCHES
				SCALE .0188 SCALE



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(B)MACH = 2.86

DATA SET SYMBOL CONFIGURATION DESCRIPTION V/MOD NOSE CHS

(DP6208) LA-BC: UPVT1040: GRBITTER 0898 V/MOD: NOSE +CHS

(DP6209) LA-BC: UPVT1040: GRBITTER 0898 V/MOD: NOSE +CHS

(DP6204) LA-BC: UPVT1040: GRBITTER 0898 V/MOD: NOSE +CHS

(DP6202) LA-BC: UPVT1040: GRBITTER 0898 V/MOD: NOSE +CHS

BETA ELEVTR AIRLRN JFLAP

3.000 -20.000 .000 -14.250

3.000 -20.000 .000 -14.250

3.000 -10.000 .000 -14.250

3.000 -14.250 .000 -14.250

REFERENCE INFORMATION SD, IN. INCHES

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LREF 8.9028 INCHES

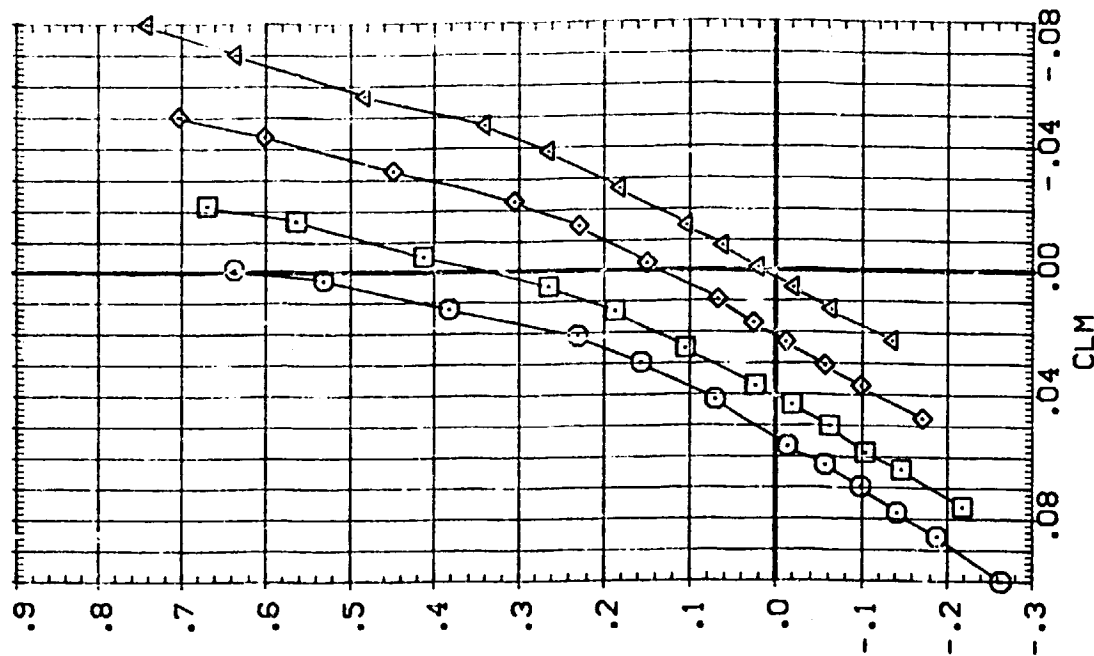
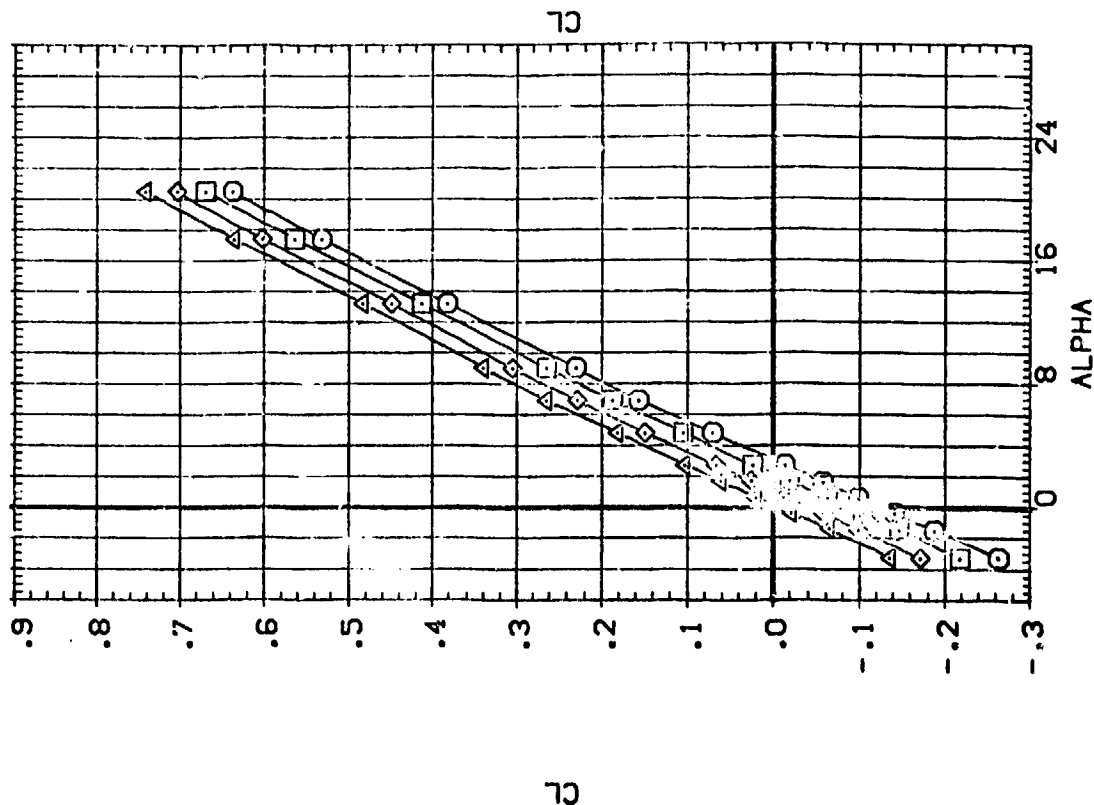
BREF 17.5628 INCHES

XWRP 15.7313 INCHES

YWRP .0000 INCHES

ZWRP .0000 INCHES

SCALE .0183 SCALE



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(MACH = 1.90)

DATA SET SYMBOL
 (DP6208)
 (DP6206)
 (DP3204)
 (DP6202)

CONFIGURATION DESCRIPTION
 LA-8C: JPV11040: CR31TER 089B V/MOD. NOSE +CMS
 LA-8C: LPVT1040: CR31TER 089B V/MOD. NOSE +CMS
 LA-8C: LPVT1040: CR31TER 089B V/MOD. NOSE +CMS
 LA-8C: LPVT1040: CR31TER 089B V/MOD. NOSE +CMS

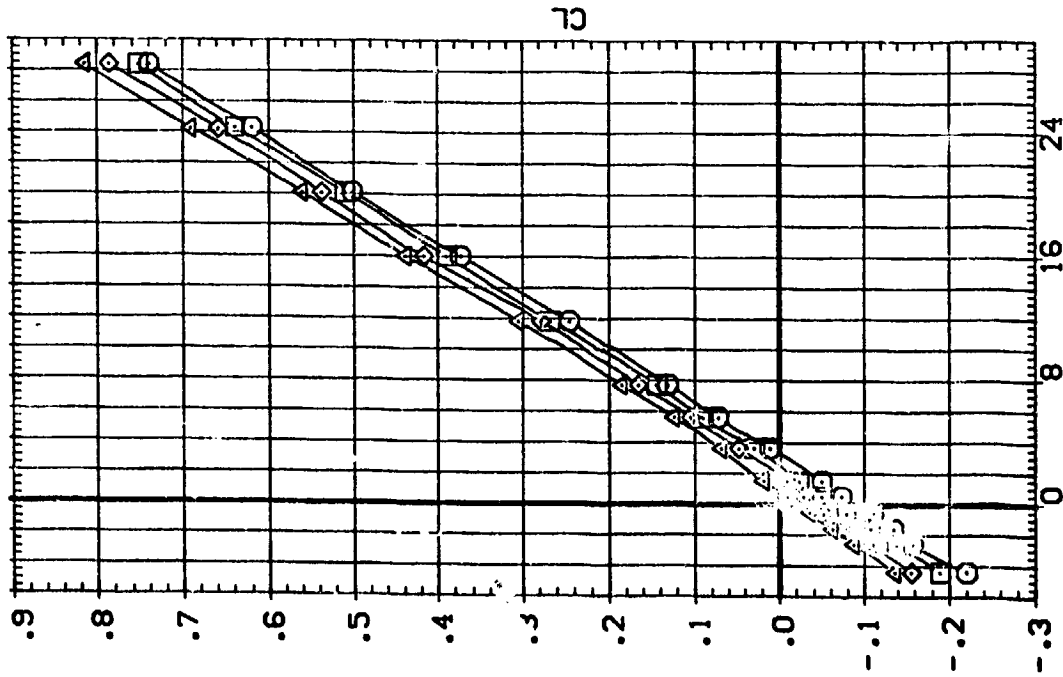
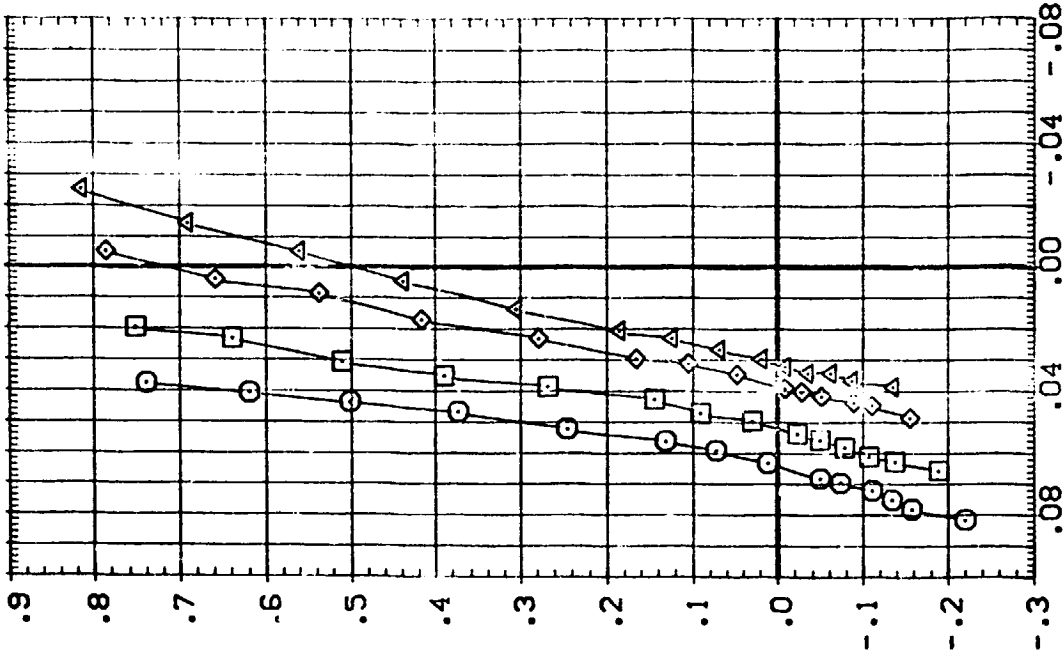
BETA
 3.000
 3.000
 3.000
 3.000

ELEVTR
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 -20.000
 -10.000

A1LRON
 .000
 .000
 .000

BOFLAP
 -14.250
 -14.250
 -14.250

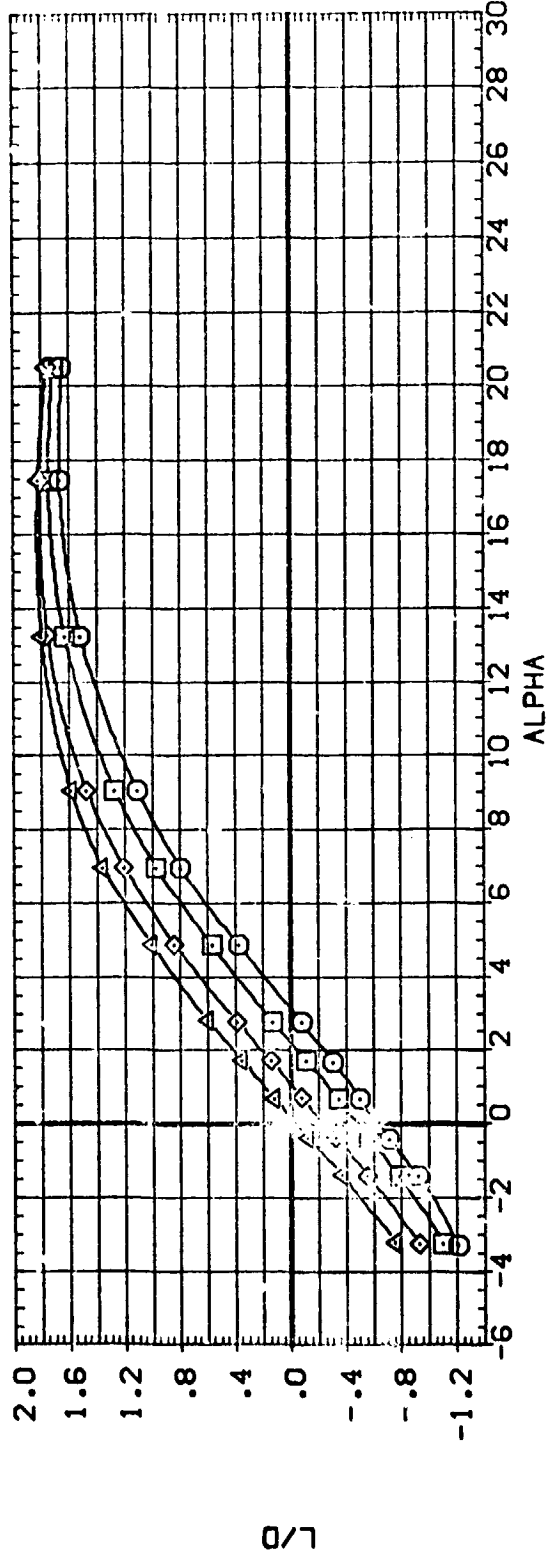
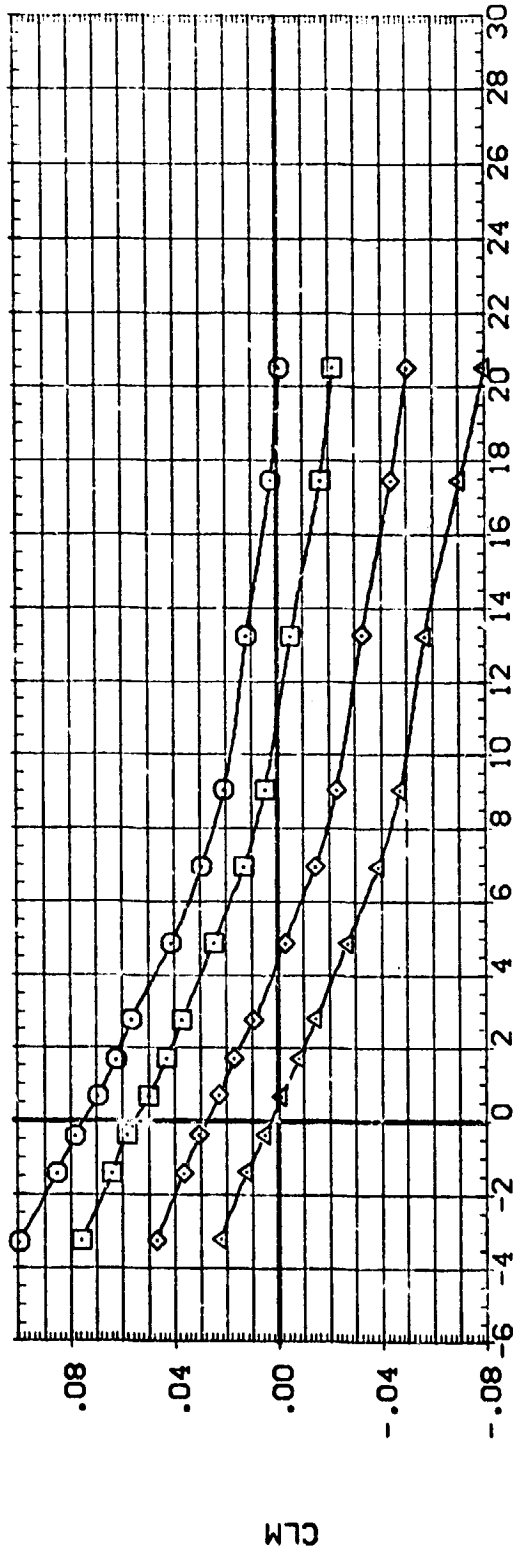
REFERENCE INFORMATION
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 LREF 8.9025 INCHES
 P-REF 17.5828 INCHES
 X1RRP 15.7313 INCHES
 Y1RRP .0000 INCHES
 Z1RRP .0000 INCHES
 SCALE .0188 SCALE



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(B)MACH = 2.86

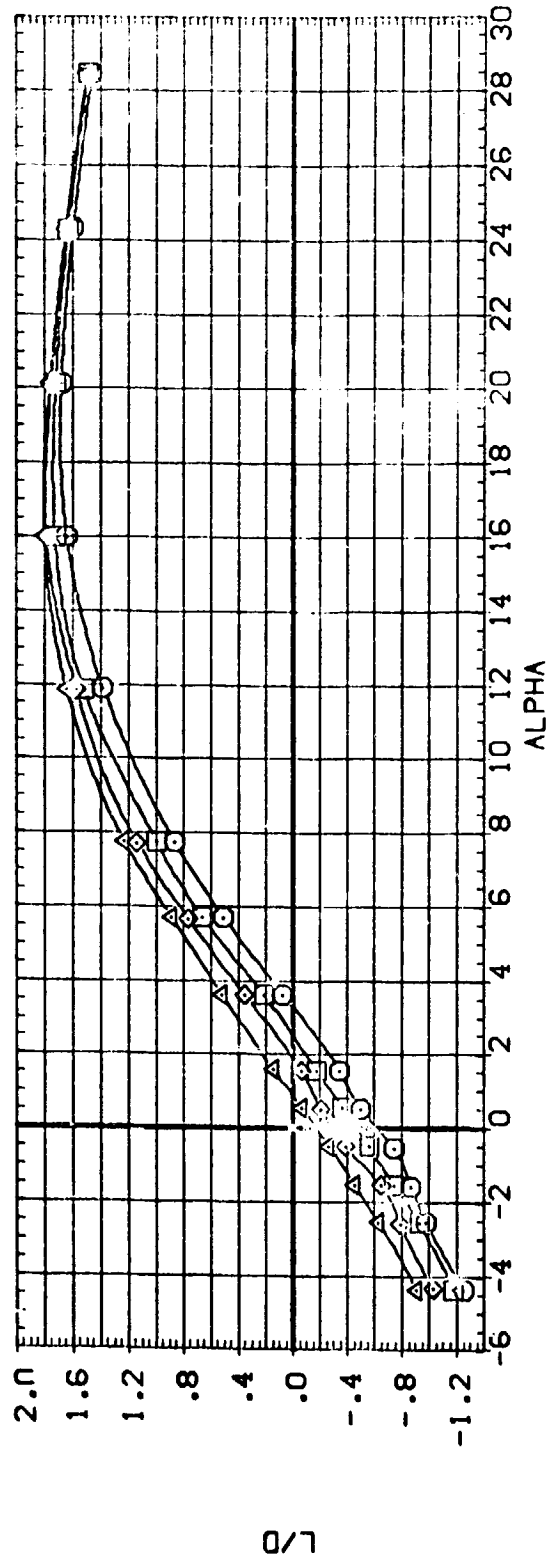
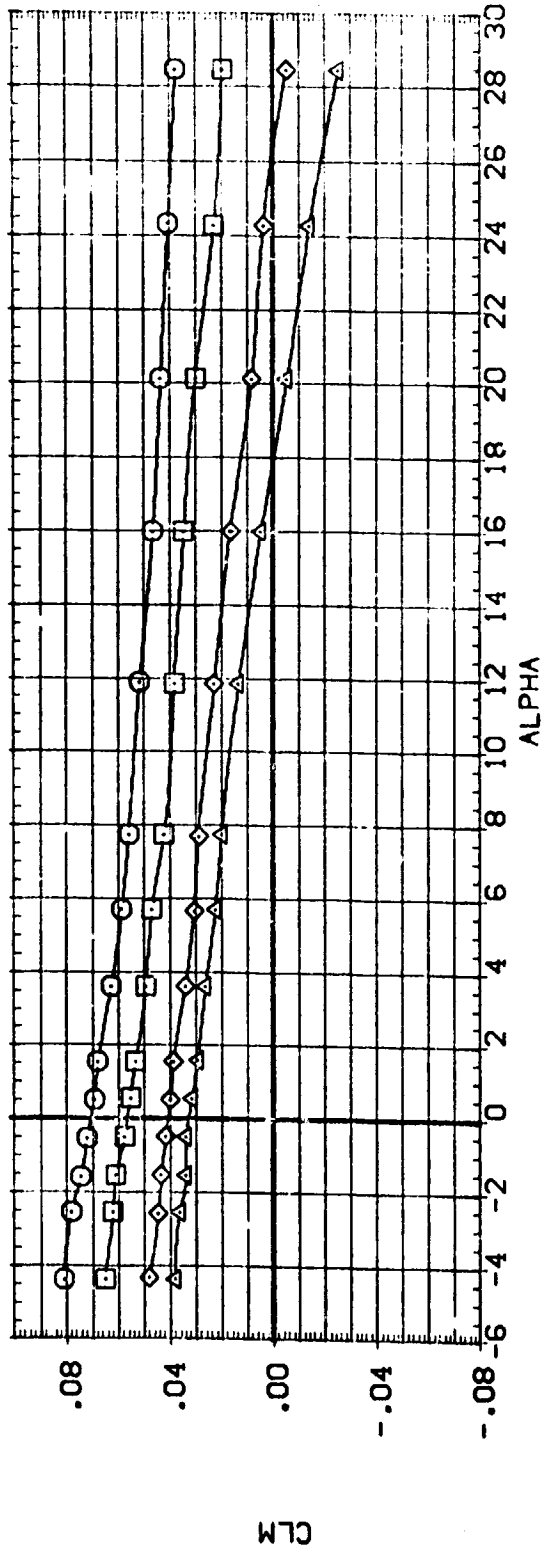
DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	ELEVTR	AILRON	BOFLAP	REFERENCE INFORMATION
(DP6208)	LA-8C: UPVT1040: CRBITER 0899 V/MOD. NOSE +0VS	3.000	-30.000	.000	-14.250	SREF 136.1808 SQ. IN.
(DP6209)	LA-8C: UPVT1040: CRBITER 0899 V/MOD. NOSE +0VS	3.000	-20.000	.000	-14.250	LREF 8.5025 INCHES
(DP6204)	LA-8C: UPVT1040: CRBITER 0899 V/MOD. NOSE +0VS	3.000	-10.000	.000	-14.250	BREF 17.5628 INCHES
(DP6202)	LA-8C: UPVT1040: CRBITER 0899 V/MOD. NOSE +0VS	3.000	.000	.000	-14.250	XVRP 15.7313 INCHES
						ZVRP .0000 INCHES
						SCALE .0168 INCHES SCALE



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

CAJMACH = 1.90

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	ELEVTR	AILRON	BOFLAP	REFERENCE INFORMATION
(DP6208)	LA-8C: UPVT1040; CRBITER 0898	3.000	-30.000	.000	-14.250	SREF 136.1808
(DP6209)	LA-8C: UPVT1040; CRBITER 0898	3.000	-20.000	.000	-14.250	LREF 9.6075
(DP6204)	LA-8C: UPVT1040; CRBITER 0898	3.000	-10.000	.000	-14.250	BREF 17.5573
(DP6202)	LA-8C: UPVT1040; CRBITER 0898	3.000	.000	.000	-14.250	XPRP 15.7313
						YMRP .0000
						ZMRP .0000
						SCALE .0168
						INCHES
						INCHES
						INCHES
						SCALE



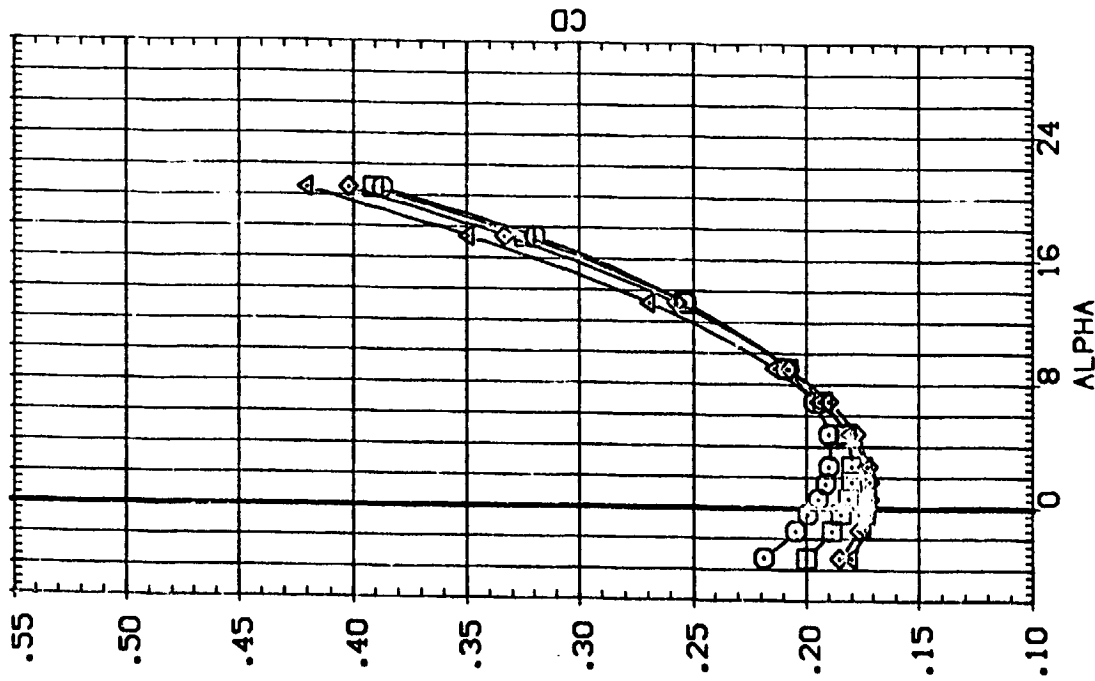
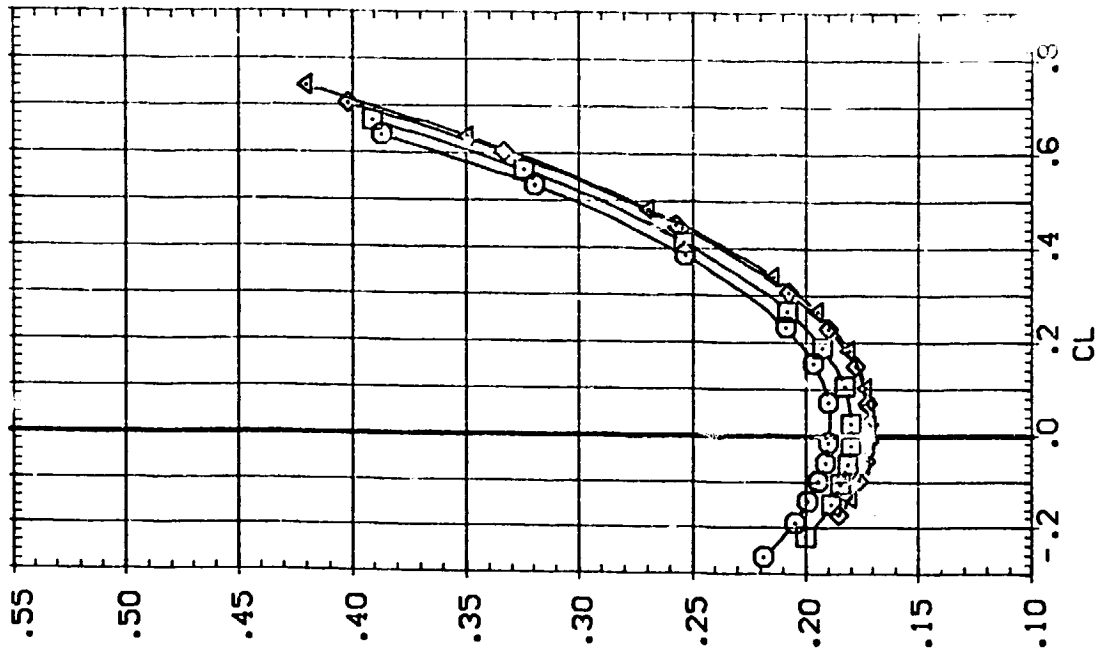
EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(B)MACH = 2.86

PAGE

8

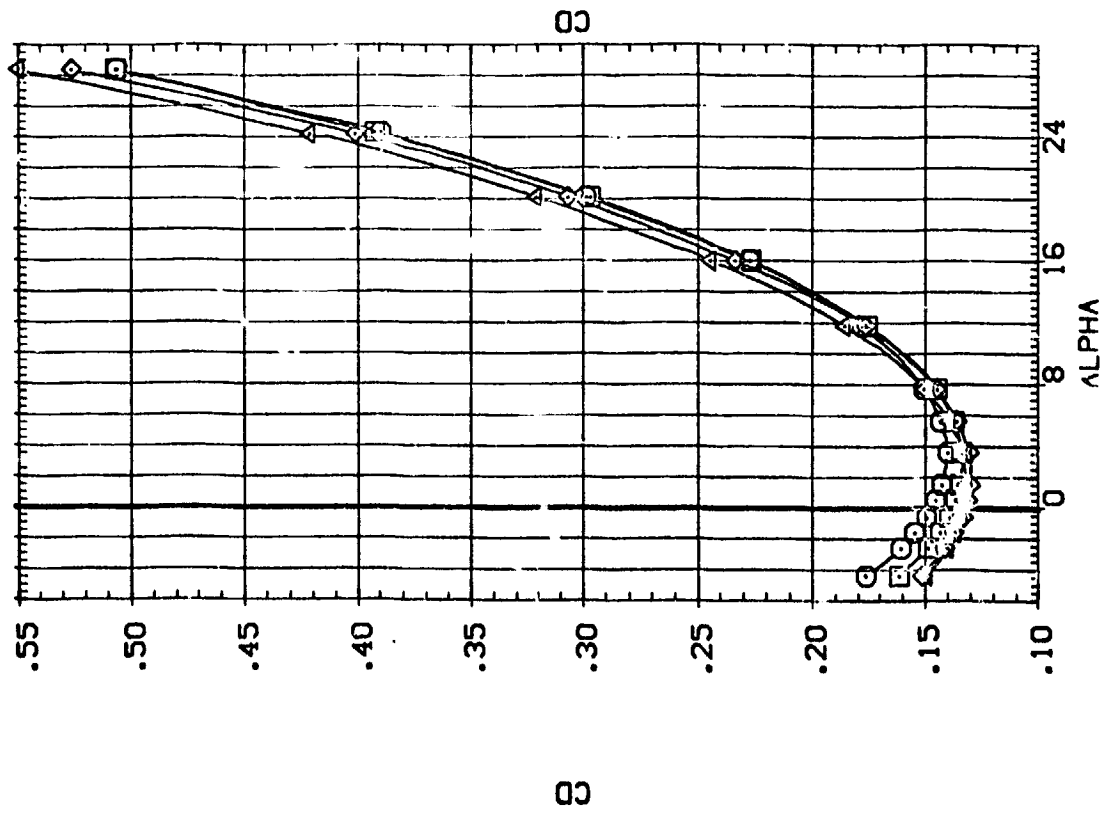
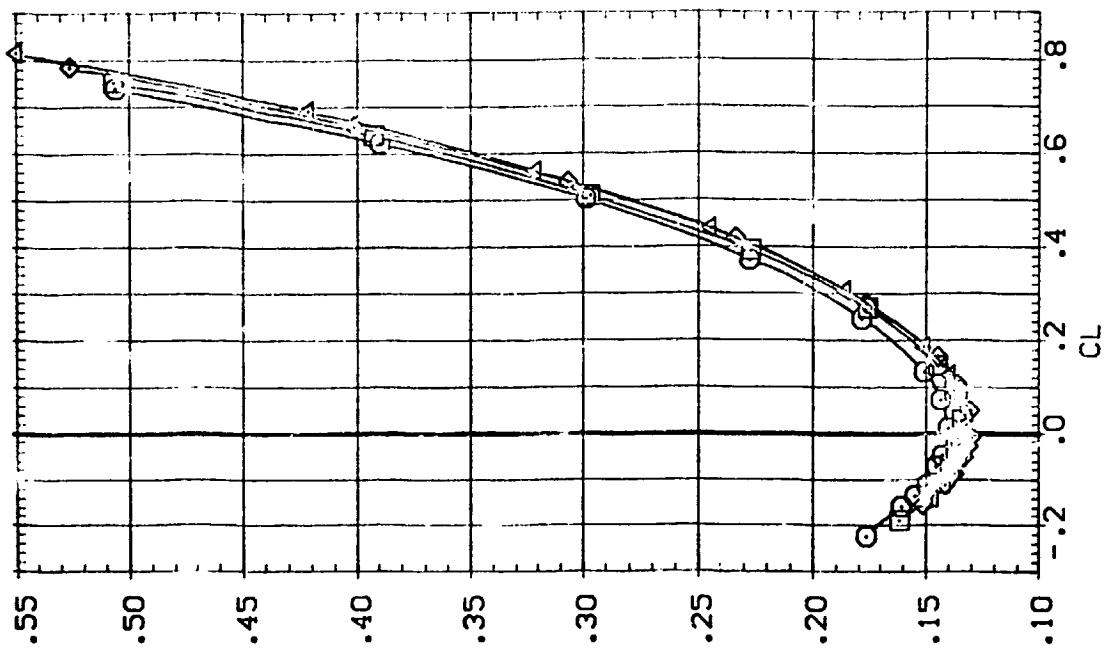
DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	ELEVTR	AILRON	BDFLAP	REFERENCE INFORMATION
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(DP6206)	LA-8C: UPVT1040; DRB1TER 0898 V/MCD; NOSE +0MS	3.000	-20.000	.000	-14.250	LREF 8.9025 INCHES
(DP6204)	LA-8C: UPVT1040; DRB1TER 0898 V/MCD; NOSE +0MS	3.000	-10.000	.000	-14.250	BREF 17.5628 INCHES
(DP6202)	LA-8C: UPVT1040; DRB1TER 0898 V/MCD; NOSE +0MS	3.000	.000	.000	-14.250	XMRP 15.7313 INCHES
						ZMRP .0000 INCHES
						SCALE .0188 INCHES



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(A) MACH = 1.90

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	ELEVTR	AIRLON	BOFLAP	REFERENCE INFORMATION	50 IN
(DP6208)	LA-8C. UPVT1040. CRBITER 0898 V/MOD. NOISE +OMS	3.000	-30.000	.000	-14.250	SREF 136.1808	INCHES
(DP6206)	LA-8C. UPVT1040. CRBITER 0898 V/MOD. NOISE +OMS	3.000	-20.000	.000	-14.250	LREF 8.9025	INCHES
(DP6204)	LA-8C. UPVT1040. CRBITER 0898 V/MOD. NOISE +OMS	3.000	-10.000	.000	-14.250	BREF 17.5628	INCHES
(DP6202)	LA-8C. UPVT1040. CRBITER 0898 V/MOD. NOISE +OMS	3.000	.000	.000	-14.250	XMRP 15.7313	INCHES
						ZMRP .0000	INCHES
						SCALE .0188	SCALE



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(B)MACH = 2.86

DATA SET SYMBOL CONFIGURATION DESC: 13N

(DP6209)	LA-8C: UPVT1040: CRBITER	0893	V/MOD: NOSE	+CMS
(DP6205)	LA-8C: UPVT1040: CRBITER	0893	V/MOD: NOSE	+CMS
(DP6203)	LA-8C: UPVT1040: CRBITER	0893	V/MOD: NOSE	+CMS
(DP6201)	LA-8C: UPVT1040: CRBITER	0893	V/MOD: NOSE	+CMS

BETA: .000
.000
.000
.000

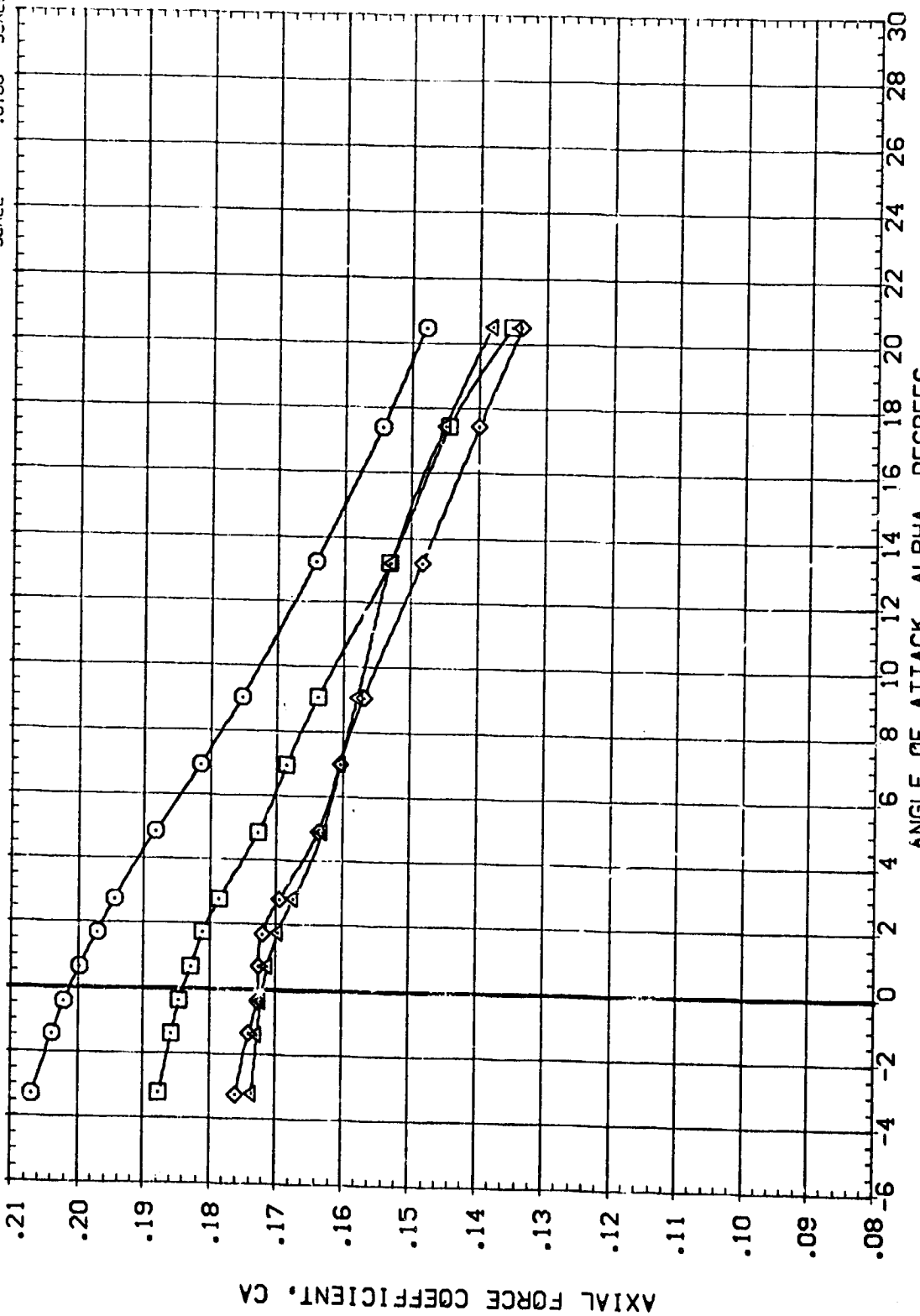
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-10.000
.000

AILRON: .000
.000
.000
.000

BOFLAP: -14.250
-14.250
-14.250
-14.250

REFERENCE INFORMATION

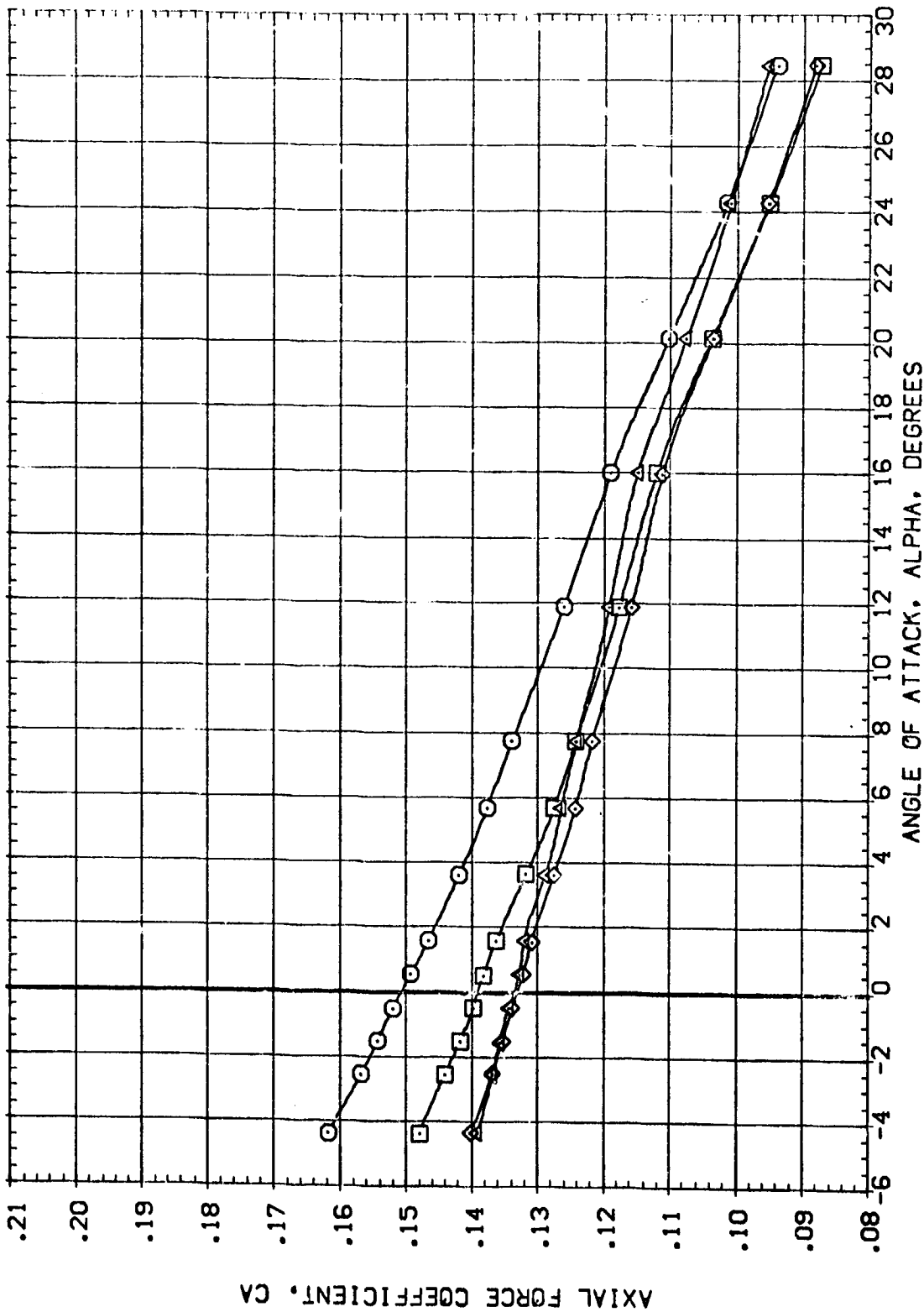
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LREF	8.9025	INCHES
BREF	17.5628	INCHES
YMRP	15.7313	INCHES
ZMRP	.0000	INCHES
SCALE	.0188	SCALE



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(M)MACH = 1.90

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (DP6209) □ LA-8C: UPVT1040: ORBITER 0898 V/MOD. NOISE +0MS
 (DP6205) ○ LA-8C: UPVT1040: ORBITER 0899 V/MOD. NOISE +0MS
 (DP6203) △ LA-8C: UPVT1040: ORBITER 0899 V/MOD. NOISE +0MS
 (DP6201) ◇ LA-8C: UPVT1040: ORBITER 0898 V/MOD. NOISE +0MS



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

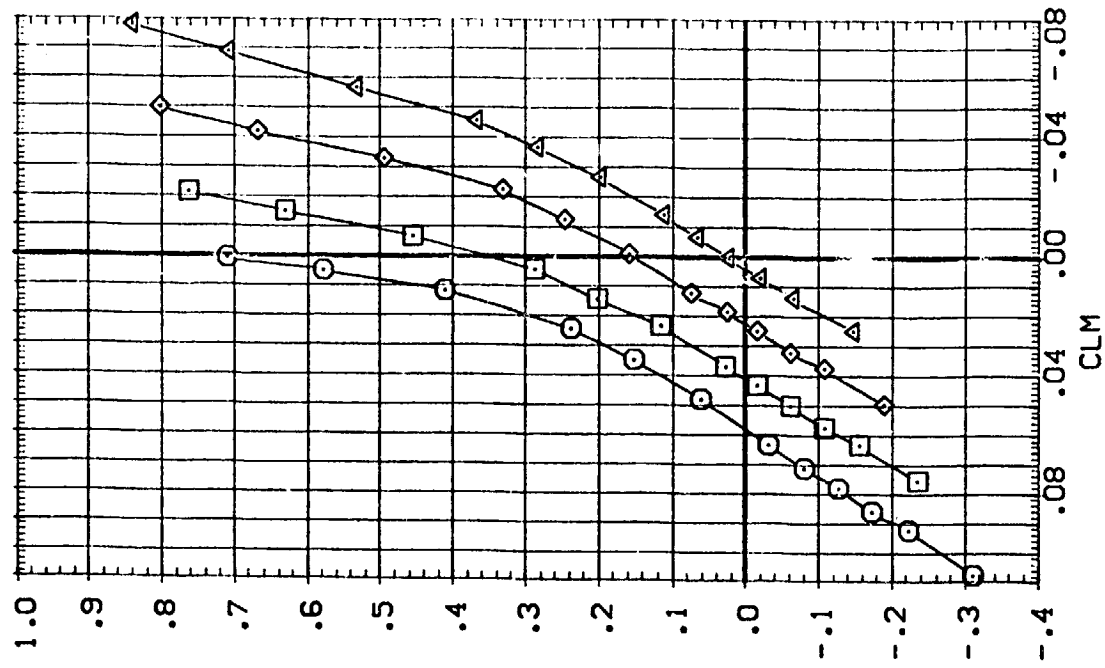
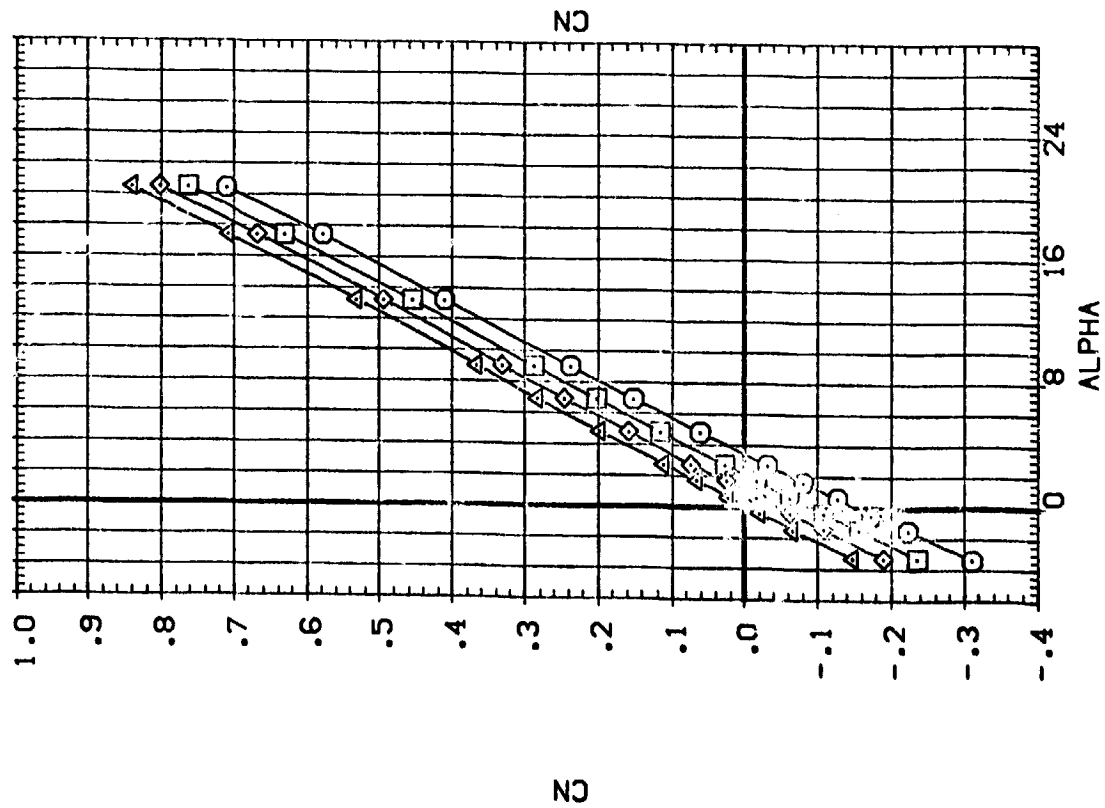
(8)MACH = 2.86

REFERENCE INFORMATION
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 BREF 17.5528 INCHES
 XMRP 1.7313 INCHES
 YMRP .0000 INCHES
 ZMRP .0000 INCHES
 SCALE .0188

DATA SET SYMBOL CONFIGURATION DESCRIPTION
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 (DP6205) ◊ LA-8C: LPVT1040: CRBITTER C899 V/MOD. NOSE +CMS
 (DP6203) ○ LA-8C: LPVT1040: CRBITTER C899 V/MOD. NOSE +CMS
 (DP6201) △ LA-8C: LPVT1040: CRBITTER C899 V/MOD. NOSE +CMS

BETA ELEVTR AILRON BOFLAP
 .000 -30.000 .000 -14.250
 .000 -20.000 .000 -14.250
 .000 -10.000 .000 -14.250

REFERENCE INFORMATION
 SREF 136.1808 SQ. IN.
 LREF 8.9025 INCHES
 BREF 17.5528 INCHES
 XWRP 15.7313 INCHES
 YWRP .0000 INCHES
 ZWRP .0000 INCHES
 SCALE .0188 SCALE



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(A)MACH = 1.90

DATA SET SYMBOL

(DP6209)
(DP6205)
(DP6203)
(DP6201)

CONFIGURATION DESCRIPTION

LA-8C: UPVT1040: ORBITER 0898 V/MOD. NOSE +OMS
 LA-8C: UPVT1040: ORBITER 0899 V/MOD. NOSE +OMS
 LA-8C: UPVT1040: ORBITER 0899 V/MOD. NOSE +OMS
 LA-8C: UPVT1040: ORBITER 0898 V/MOD. NOSE +OMS

BETA
 .000
 .000
 .000
 .000
 .000

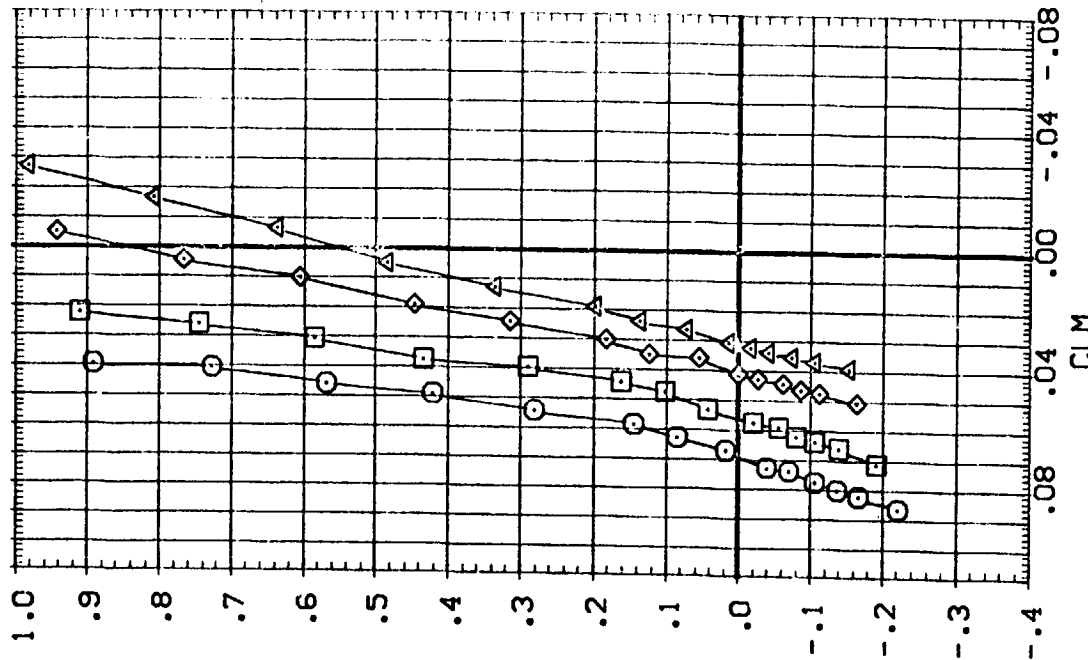
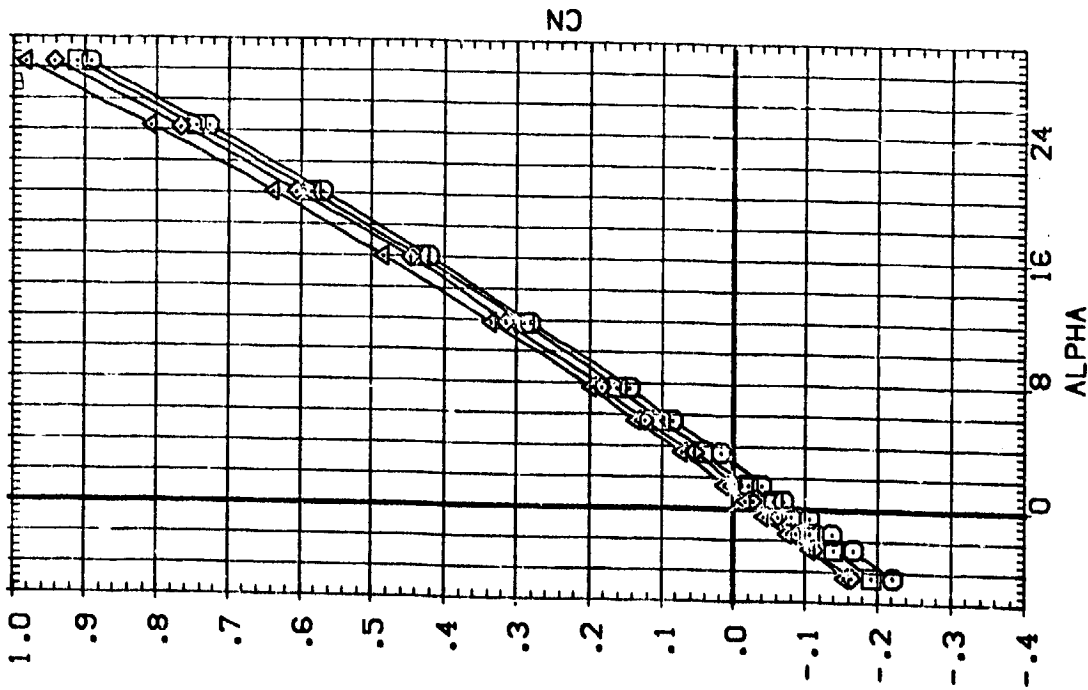
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 .000

AILRON
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 .000
 .000
 .000
 .000

BOFLAP
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 -14.250
 -14.250
 -14.250
 -14.250

REFERENCE INFORMATION

SREF 136.1808 SQ. IN. INCHES
 LREF 8.5025 INCHES
 BREF 17.5528 INCHES
 XMRP 15.7313 INCHES
 YMRP .0000 INCHES
 ZMRP .0000 INCHES
 SCALE .0188 SCALE



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

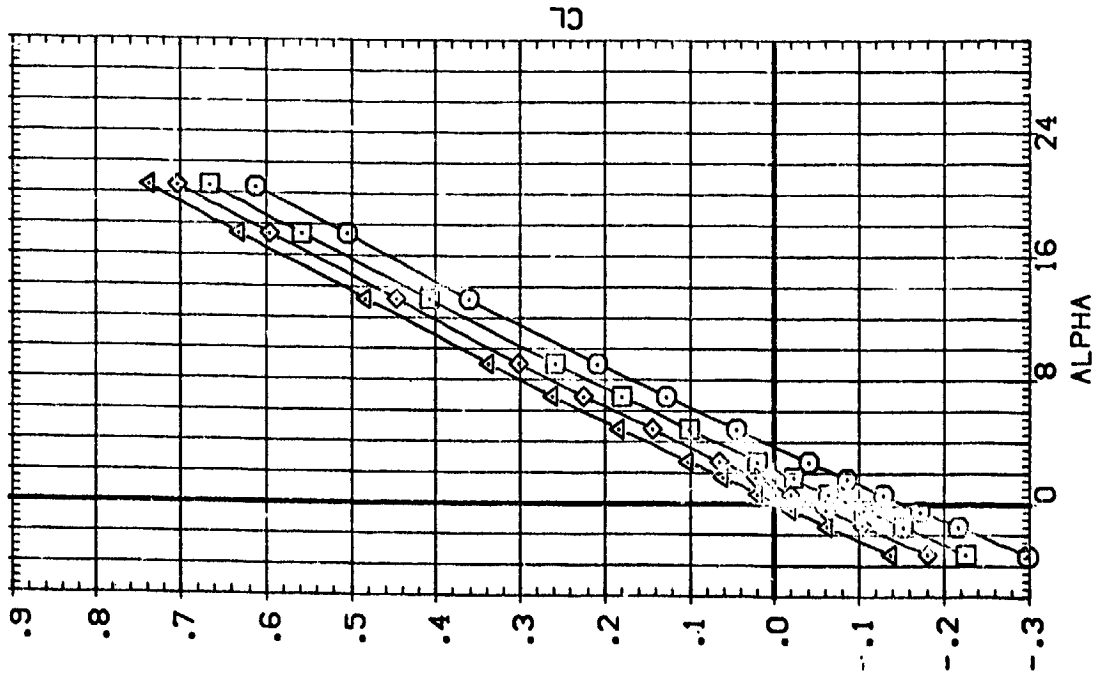
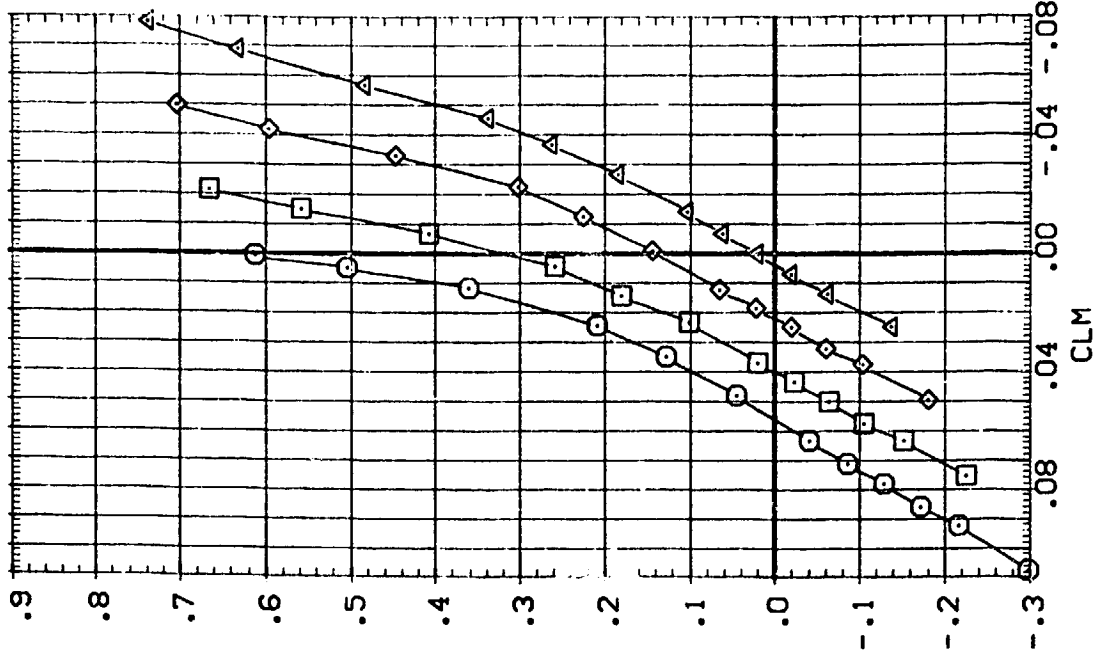
(B)MACH = 2.86

DATA SET SYMBOL CONFIGURATION DESCRIPTION REFERENCE INFORMATION

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(DP6205)	LA-8C: LPVT1040: ORBITTER DB99 V/MOD: NOSE +ONS	LREF 8.9025 INCHES
(DP6203)	LA-8C: LPVT1040: ORBITTER DB99 V/MOD: NOSE +ONS	BREF 17.5628 INCHES
(DP6201)	LA-8C: LPVT1040: ORBITTER DB99 V/MOD: NOSE +ONS	XWRP 15.7313 INCHES
		ZWRP .0000 INCHES
		SCALE .0168 INCHES

BETA ELEVTR ALLRON BOFLAP SREF SQ. IN.

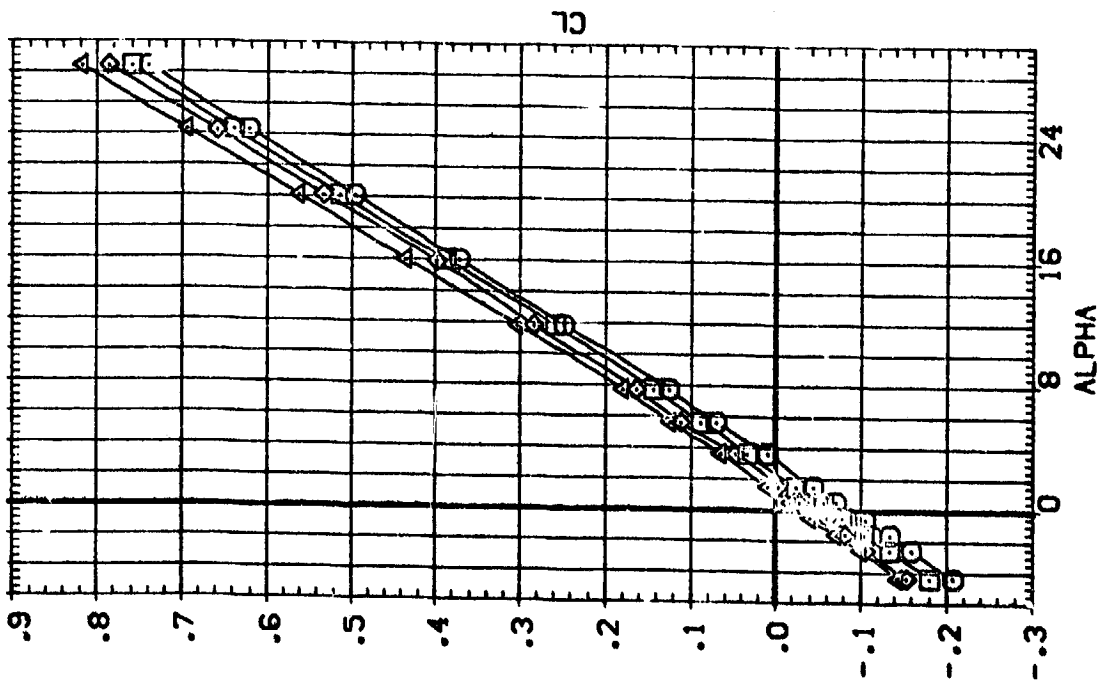
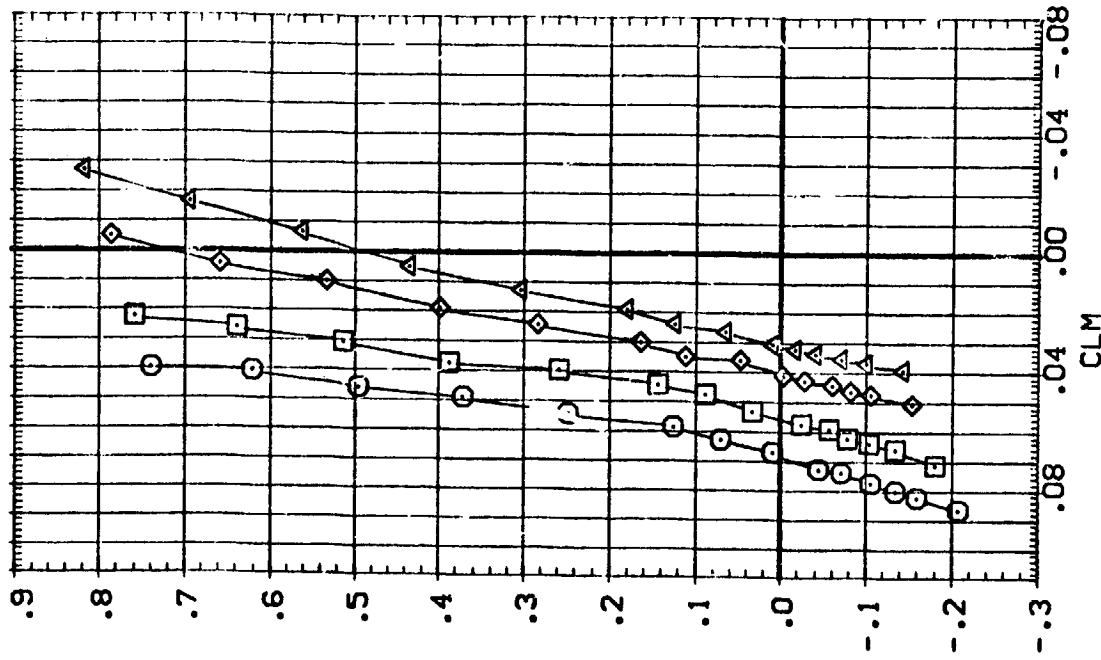
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.000	.000	.000	-14.250	15.7313	
				.0000	
				.0168	



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(A) MACH = 1.90

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	ELEVTR	AILRON	BOFLAP	REFERENCE INFORMATION
(DP6208)	LA-8C. UPVT1040. CRBITER 0898 V/MOD. NOISE +0MS	.000	-30.000	.000	-14.250	SREF 136.1808 SO.IN.
(DP6205)	LA-8C. UPVT1040. CRBITER 0898 V/MOD. NOISE +0MS	.000	-20.000	.000	-14.250	LREF 8.9025 INCHES
(DP6203)	LA-8C. UPVT1040. CRBITER 0898 V/MOD. NOISE +0MS	.000	-10.000	.000	-14.250	BREF 17.5628 INCHES
(DP6201)	LA-8C. UPVT1040. CRBITER 0898 V/MOD. NOISE +0MS	.000	.000	.000	-14.250	YMRF 15.7313 INCHES
						ZMRP .0000 INCHES
						ZMRP .0000 INCHES
						SCALE .0188



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(B)MACH = 2.86

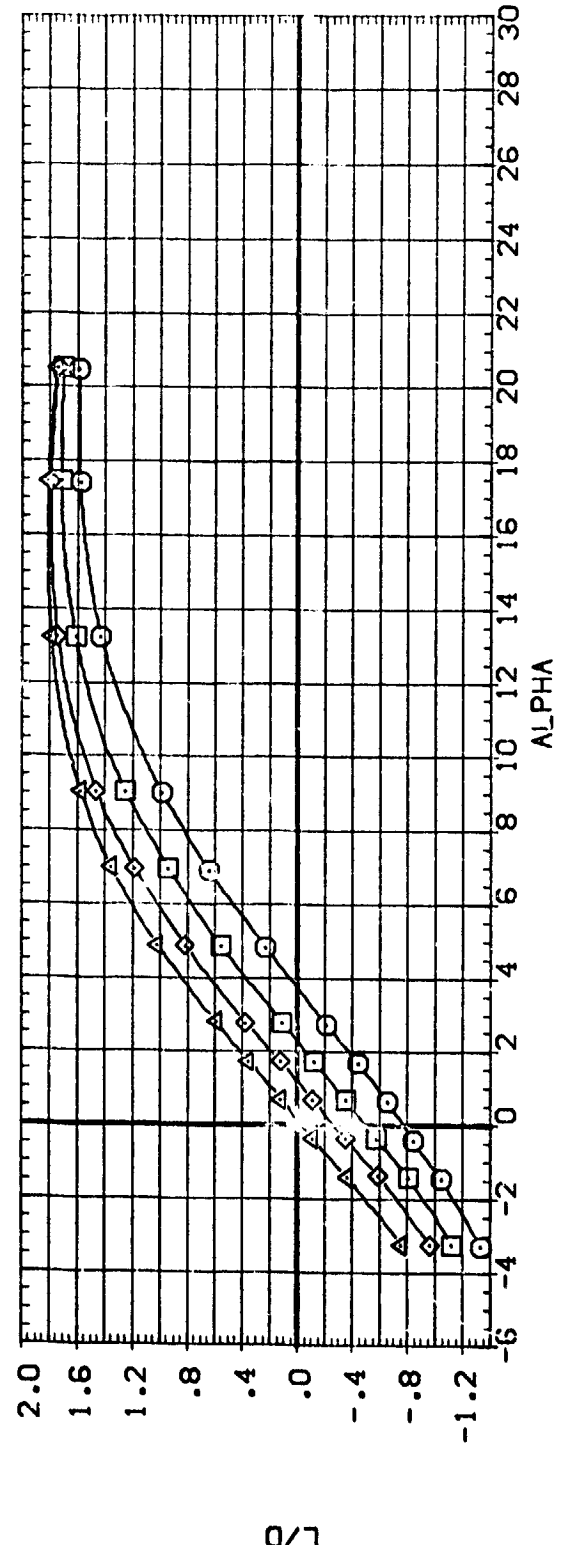
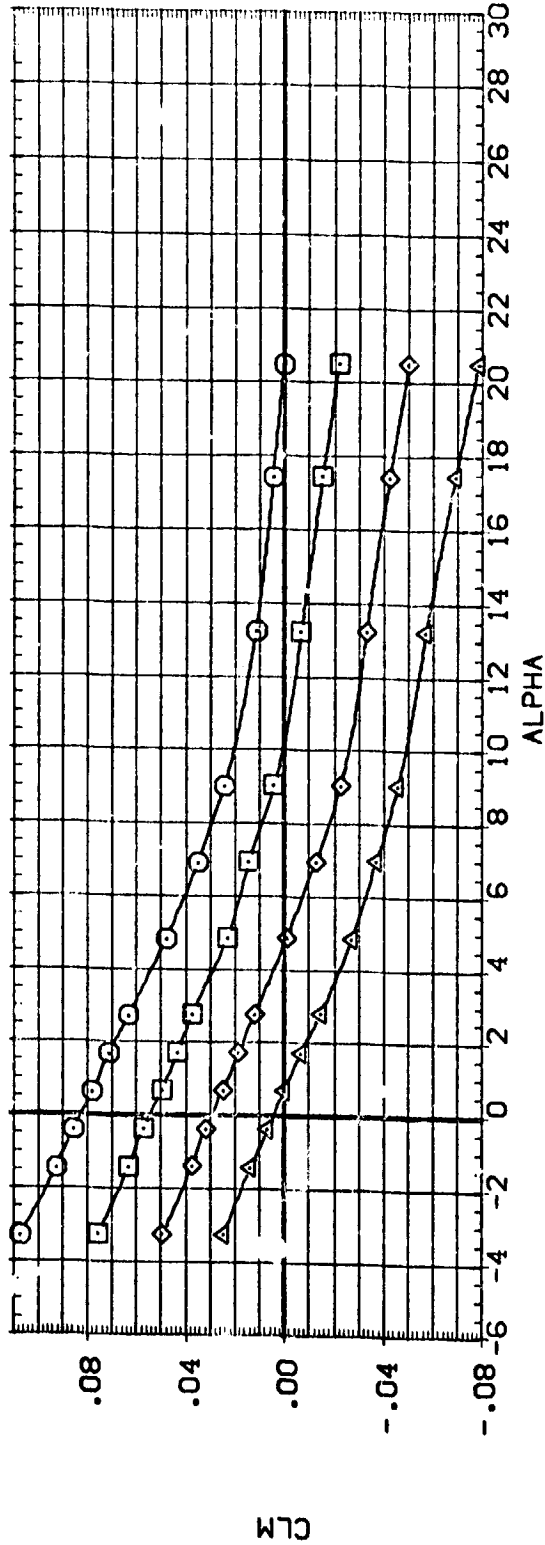
DATA SET SYMBOL
 (DP6209) □
 (DP6205) ○
 (DP6203) ◇
 (DP6201) △

CONFIGURATION DESCRIPTION
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 LA-8C: UPVT1040, CRBITER 0893
 LA-8C: UPVT1040, CRBITER 0898
 LA-8C: UPVT1040, CRBITER 0898

V/MOD. NOISE +ONS
 V/MOD. NOISE +ONS
 V/MOD. NOISE +ONS
 V/MOD. NOISE +ONS

BETA ELEVTR AIRLON BOFLAP
 .000 -30.000 .000 -14.250
 .000 -20.000 .000 -14.250
 .000 -10.000 .000 -14.250
 .000 .000 .000 .000

REFERENCE INFORMATION
 SREF 136.1808 SQ. IN.
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 SCALE .0189 SCALE

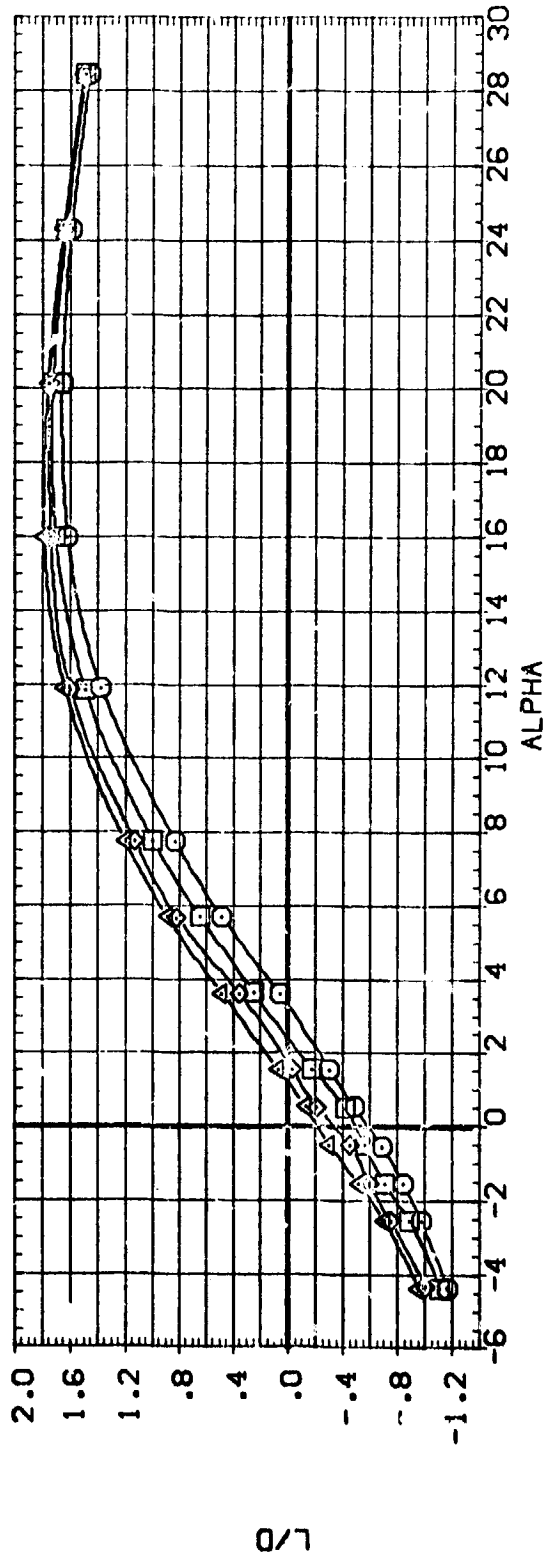
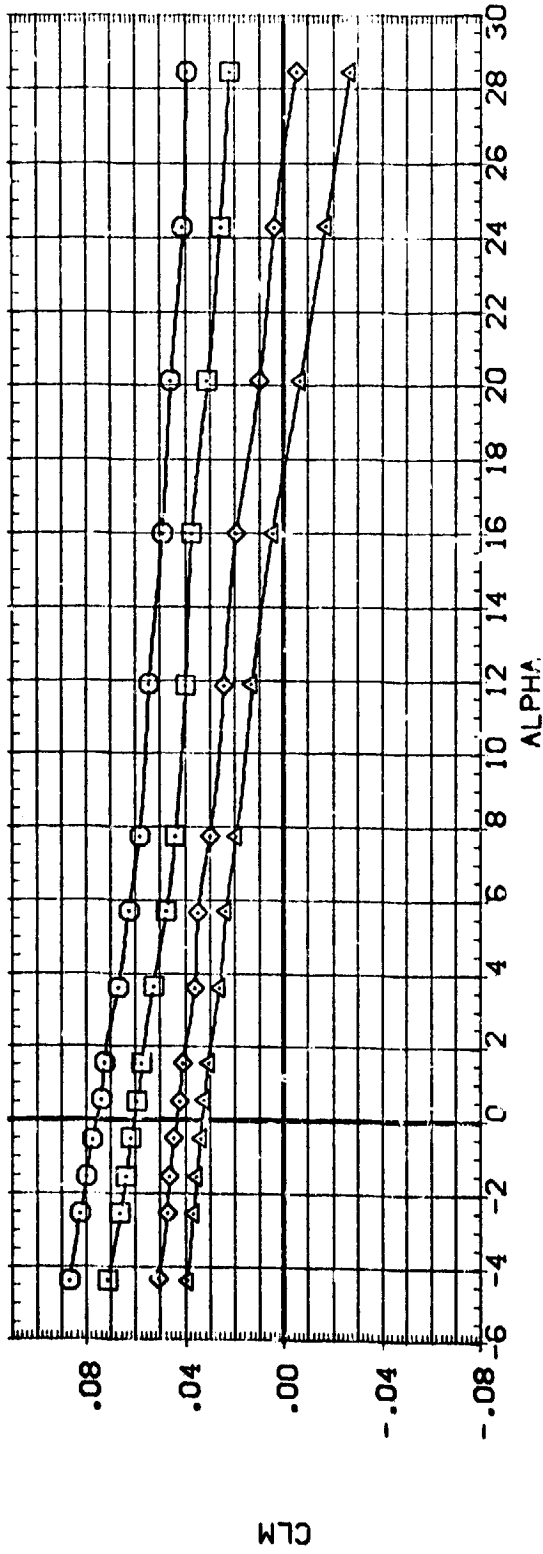


EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(A)MACH = 1.90

DATA SET SYMBOL CONFIGURATION DESCRIPTION BETA ELEVTR ATLRGN BDFLAP REFERENCE INFORMATION

DATA SET SYMBOL	CONFIGURATION DESCRIPTION	BETA	ELEVTR	ATLRGN	BDFLAP	REFERENCE INFORMATION
(DP6209)	LA-8C: UPVT1040. ORBITER 0898 W/MOD. NOSE +0MS	.000	-30.000	.000	-14.250	SREF 136.1808 SO. IN.
(DP6205)	LA-8C: UPVT1040. ORBITER 0898 W/MOD. NOSE +0MS	.000	-20.000	.000	-14.250	LREF 8.9025 INCHES
(DP6203)	LA-8C: UPVT1040. ORBITER 0893 W/MOD. NOSE +0MS	.000	-10.000	.000	-14.250	BREF 17.5628 INCHES
(DP6201)	LA-8C: UPVT1040. ORBITER 0898 W/MOD. NOSE +0MS	.000	.000	.000	-14.250	XMRP 15.7313 INCHES
						ZMRP .0000 INCHES
						SCALE .0188 INCHES



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(B)MACH = 2.86

DATA SET SYMBOL CONFIGURATION DESCRIPTION

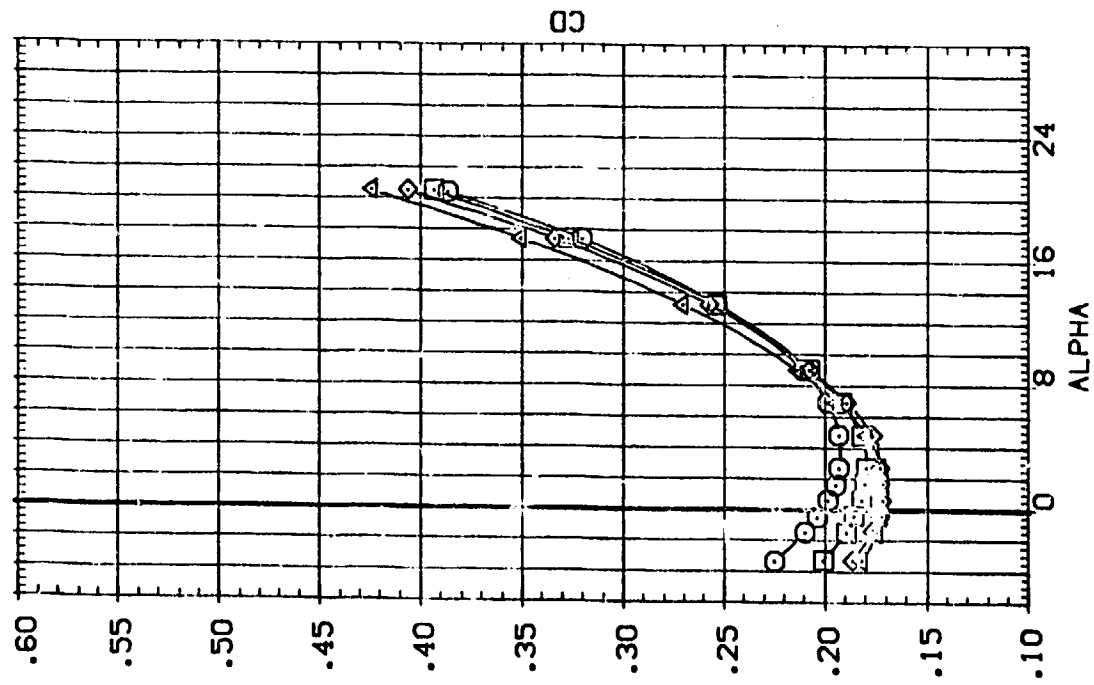
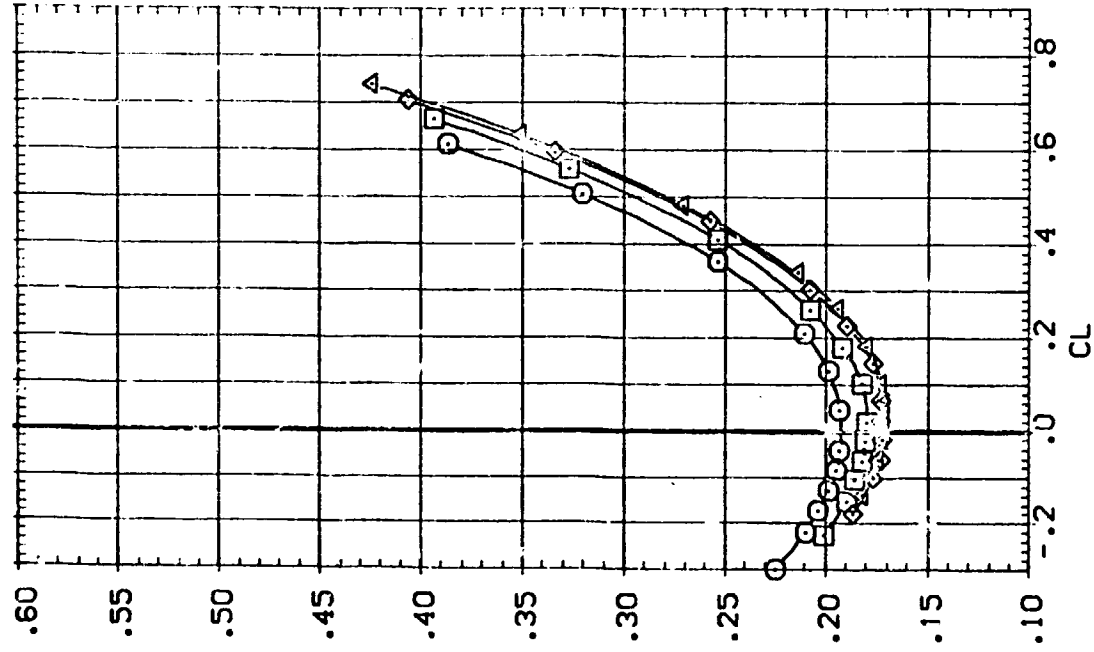
LA-8C: UPVT1040. CRBITER C899 V/MCD. NOSE +CMS
 LA-8C: UPVT1040. CRBITER C899 V/MCD. NOSE +CMS
 LA-8C: UPVT1040. CRBITER C899 V/MCD. NOSE +CMS
 LA-8C: UPVT1040. CRBITER C899 V/MCD. NOSE +CMS

BETA ELEVTR AILRON BOFLAP

.000 -30.000 .000 -14.250
 .000 -20.000 .000 -14.250
 .000 -10.000 .000 -14.250
 .000 .000 .000 -14.250

REFERENCE INFORMATION

SREF 136 1808 90. IN.
 LREF 8 3025 INCHES
 BREF 17.5628 INCHES
 XMRP 15.7313 INCHES
 ZMRP .0000 INCHES
 SCALE .0188 SCALE



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

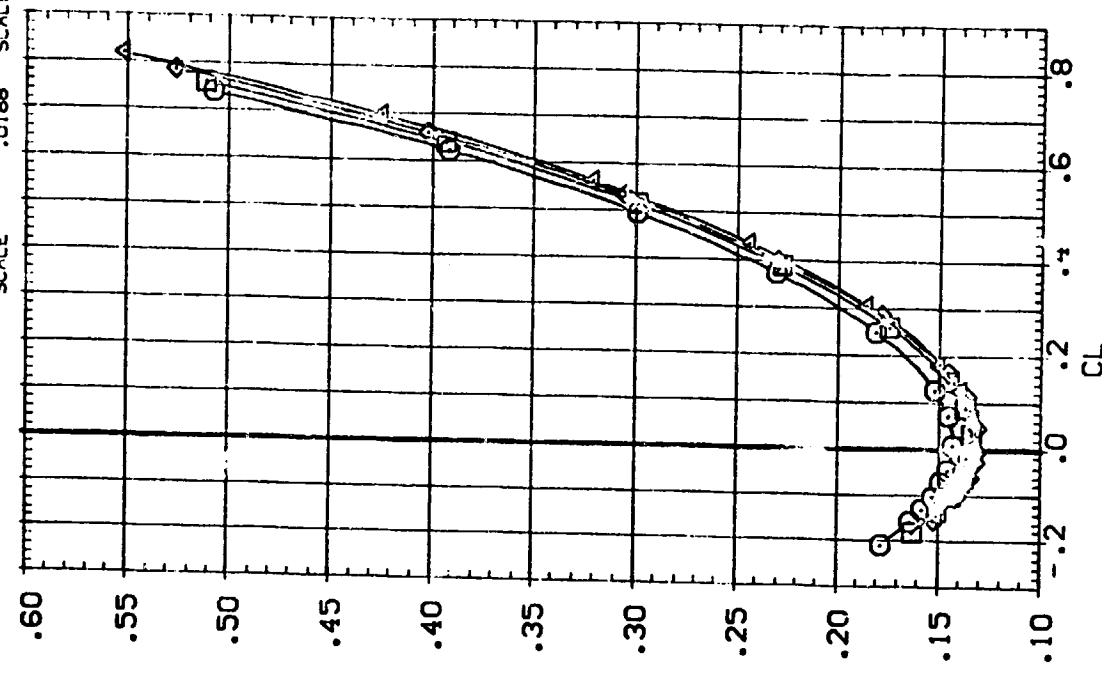
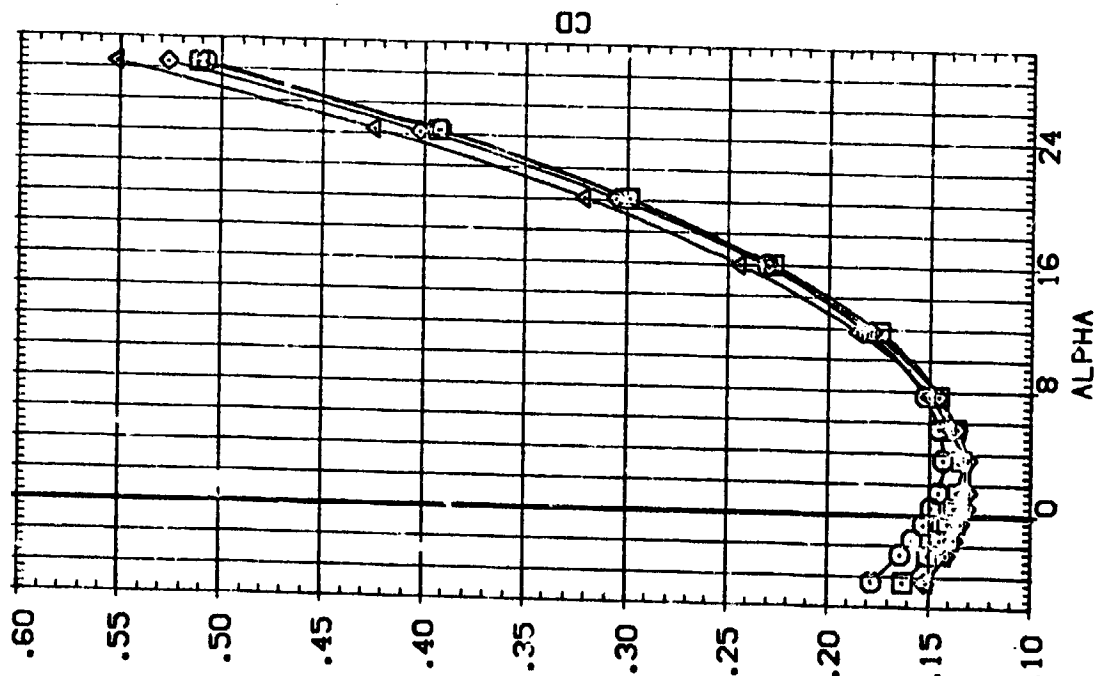
(A)MACH = 1.90

DATA SET SYMBO. (D-6209) (D-6209) (D-6209) (D-6209)

CONFIGURATION DESCRIPTION
 LA-8C: UPVT1040. ORBITER 0898 V/MOD. NOISE +CMS
 LA-8C: UPVT1040. ORBITER 0898 V/MOD. NOISE +CMS
 LA-8C: UPVT1040. ORBITER 0898 V/MOD. NOISE +CMS
 LA-8C: UPVT1040. ORBITER 0898 V/MOD. NOISE +CMS

BETA .000
 ELEVTR -30.000
 AILRON .000
 BOFLAP -14.250

REFERENCE INFORMATION
 SREF 136.1808 SO.IN.
 LREF 8.9025 INCHES
 BREF 17.5528 INCHES
 XMRP 15.7313 INCHES
 YMRP .0000 INCHES
 ZMRP .0000 INCHES
 SCALE .0188 SCALE



EFFECT OF ELEVON DEFLECTION ON LONGITUDINAL CHARACTERISTICS

(B)MACH = 2.86

DATA SET SYMBOL CONFIGURATION DESCRIPTION

(LP6C08) LA-8C: UPVT1040: ORBITER 0898 V/MOD. NOSE +OMS

(LP6C06) LA-8C: UPVT1040: ORBITER 0898 V/MOD. NOSE +OMS

(LP6C04) LA-8C: UPVT1040: CRBITER 0893 V/MOD. NOSE +OMS

(LP6C02) LA-8C: UPVT1040: CRBITER 0893 V/MOD. NOSE +OMS

ELEVTR ALLRON RUDDER

-30.000 .000

-20.000 .000

-10.000 .000

.000 .000

REFERENCE INFORMATION

SREF 136.1808 SQ. IN.

LREF 8.9025 INCHES

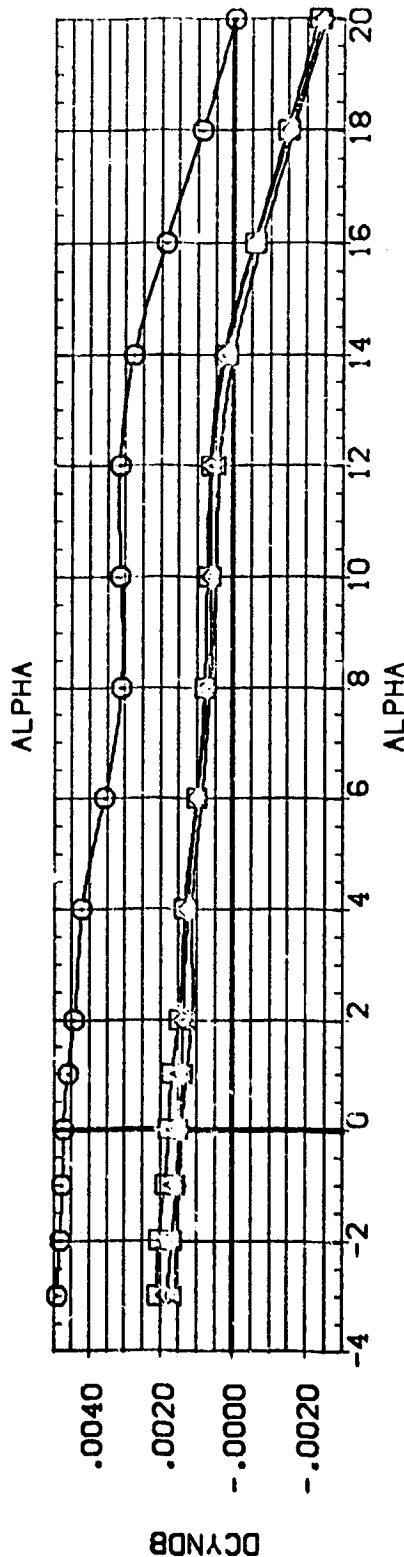
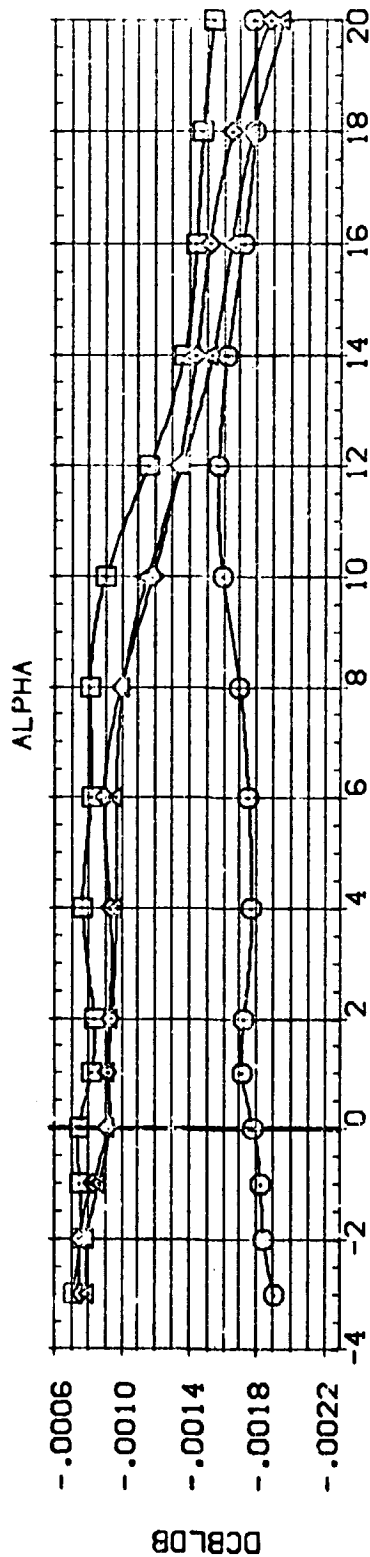
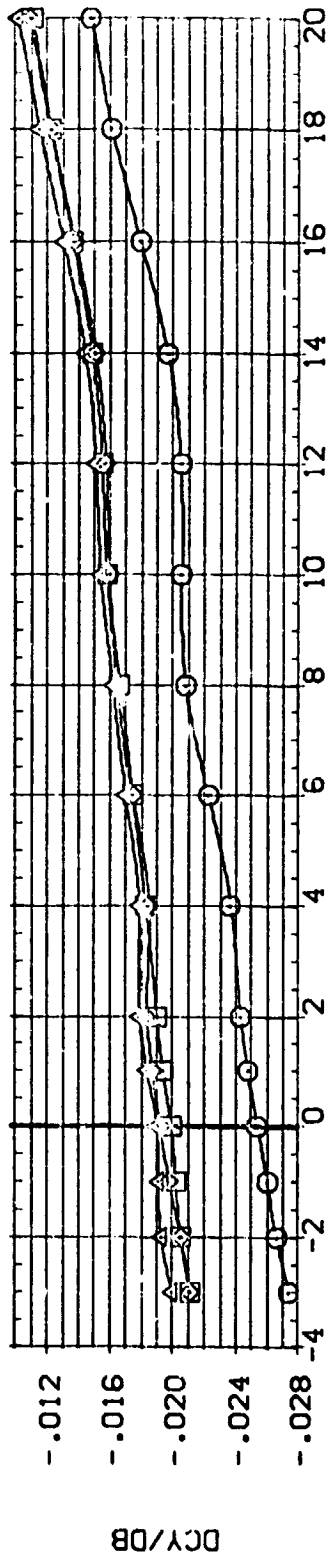
BREF 17.5628 INCHES

XMRRP 15.7313 INCHES

YMRRP .0000 INCHES

ZMRRP .0000 INCHES

SCALE .0189



EFFECT OF ELEVON DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS

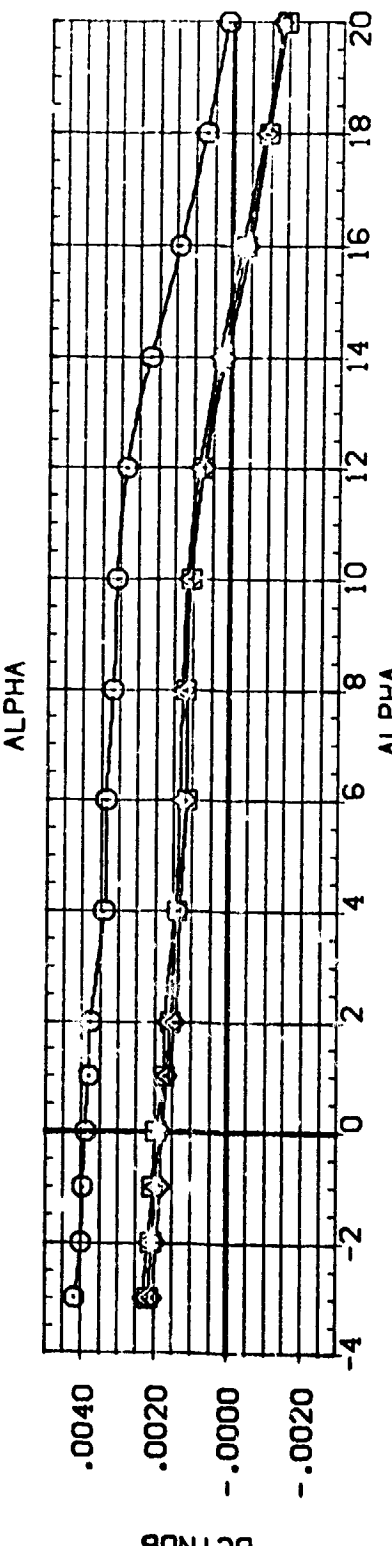
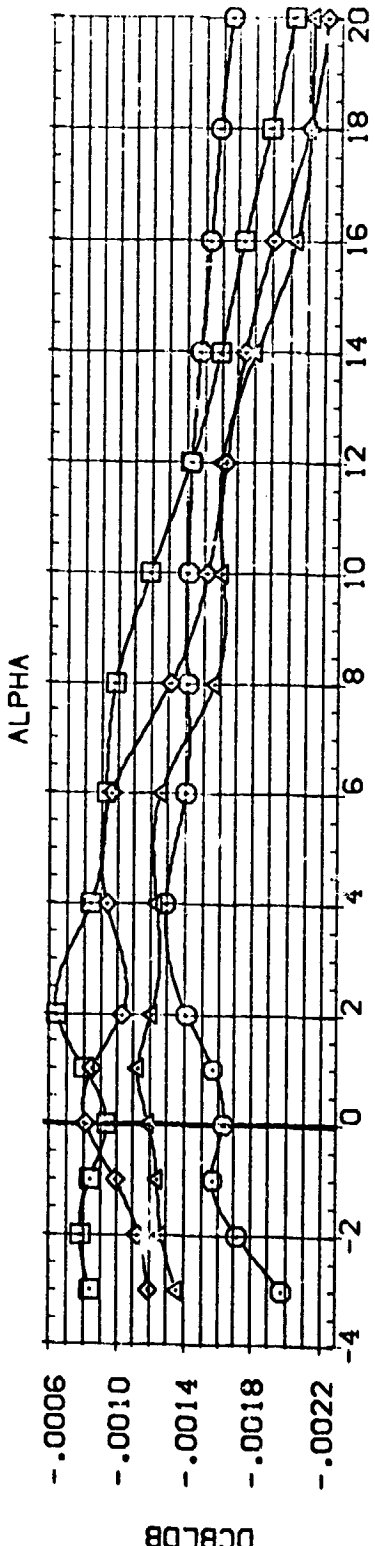
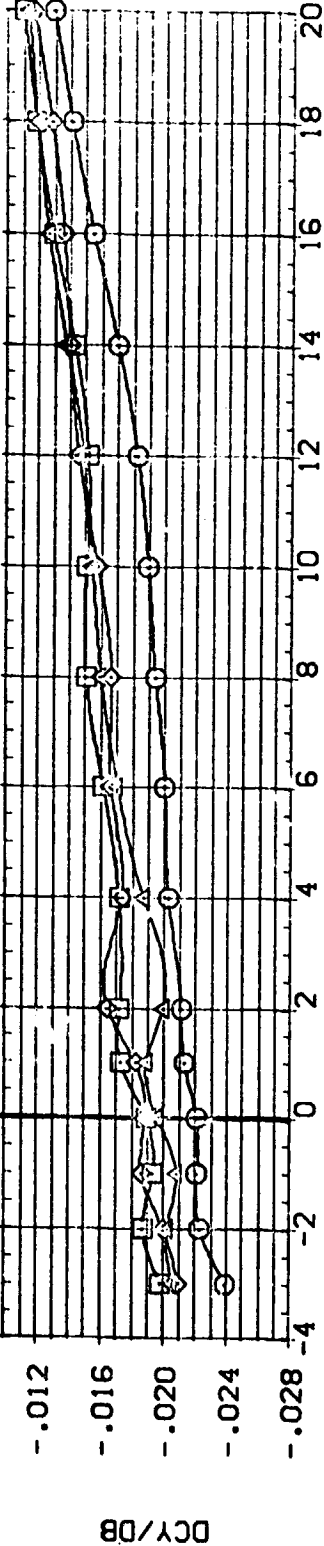
(A)MACH = 1.90

DATA SET SYMBOL CONFIGURATION DESCRIPTION REFERENCE INFORMATION

[Symbol]	LA-8C: UPVT1040; DRB1TER C898 V/MOD: NOSE +DMS	SREF	136.1808	SO. IN.	50.1
[Symbol]	LA-8C: UPVT1040; DRB1TER C898 V/MOD: NOSE +DMS	LREF	8.9023	NO. IN.	50.1
[Symbol]	LA-8C: UPVT1040; DRB1TER C898 V/MOD: NOSE +DMS	BREF	17.5528	NO. IN.	50.1
[Symbol]	LA-8C: UPVT1040; DRB1TER C898 V/MOD: NOSE +DMS	YMRP	15.7373	NO. IN.	50.1
[Symbol]	LA-8C: UPVT1040; DRB1TER C898 V/MOD: NOSE +DMS	ZMRP	.0000	NO. IN.	50.1
[Symbol]	LA-8C: UPVT1040; DRB1TER C898 V/MOD: NOSE +DMS	SCALE	.0188	SCALE	50.1

ELEVTR AIRLON RUDDER

-30.000	.000	.000
-20.000	.000	.000
-10.000	.000	.000



EFFECT OF ELEVON DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS

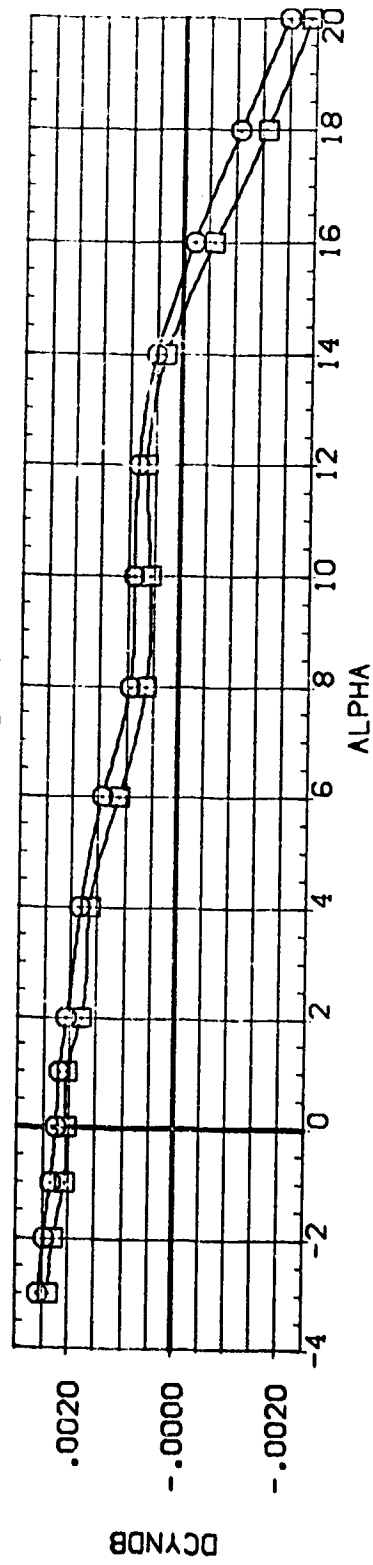
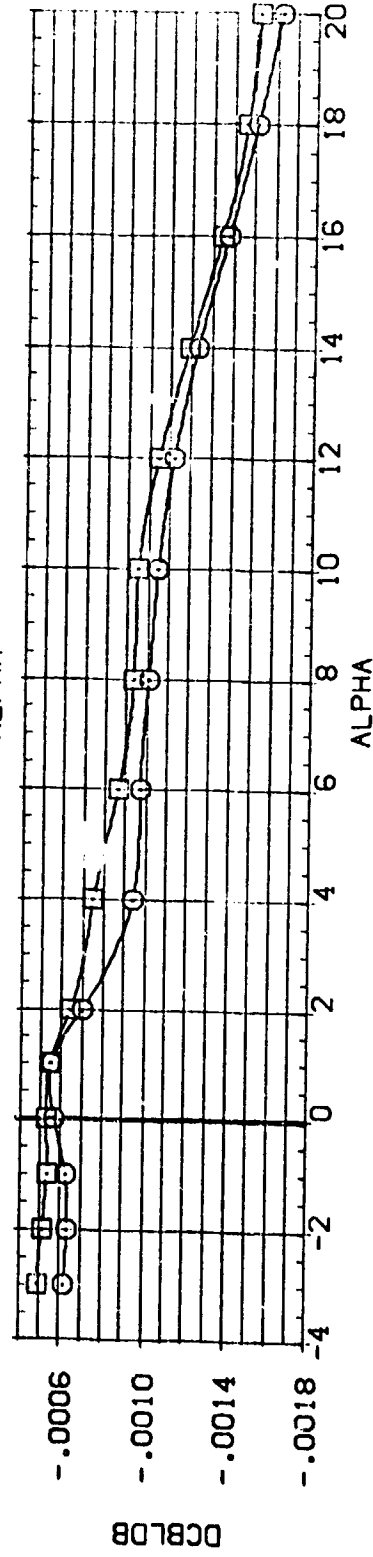
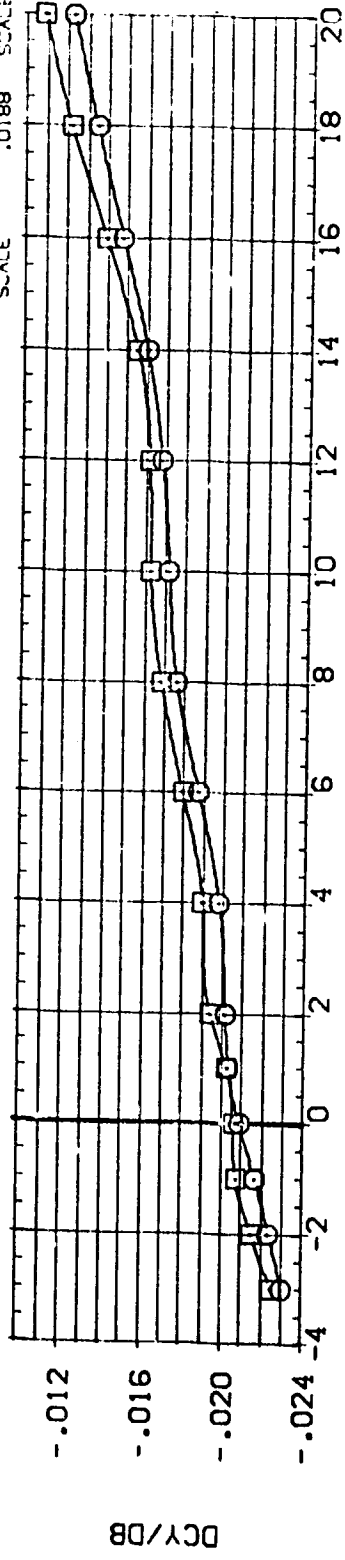
(B)MACH = 2.86

DATA SET SYMBOL
(MPC510)
(MPC508)

CONFIGURATION DESCRIPTION
LA-8C: LPVT1040. CRBTTER 089B V/MOD. NOSE +0VS
LA-8C: LPVT1040. CRBTTER 089B V/MOD. NOSE +0VS

ELEVTR AIRLON RUDDER
-30.000 .000 10.000
-30.000 .000 .000

REFERENCE INFORMATION
SREF 136.1808 SQ. IN.
LREF 8.9226 INCHES
BREF 17.5528 INCHES
XMRP 15.7313 INCHES
YMRP .0000 INCHES
ZMRP .0000 INCHES
SCALE .0188 SCALE



EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL DERIVATIVES

(M)MACH = 1.90

DATA SET SYMBOL
(MPC10)
(MPC08)

CONFIGURATION DESCRIPTION

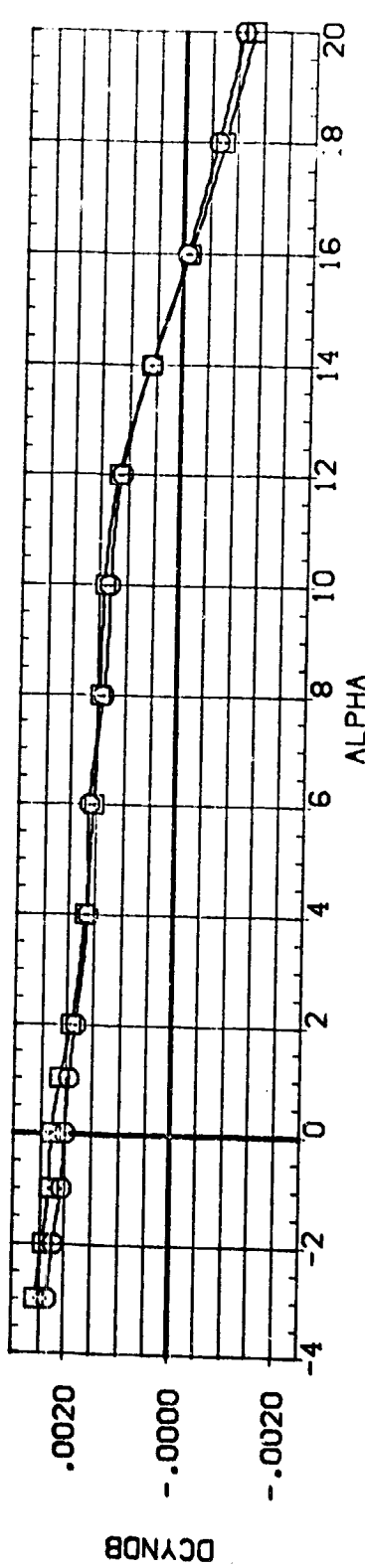
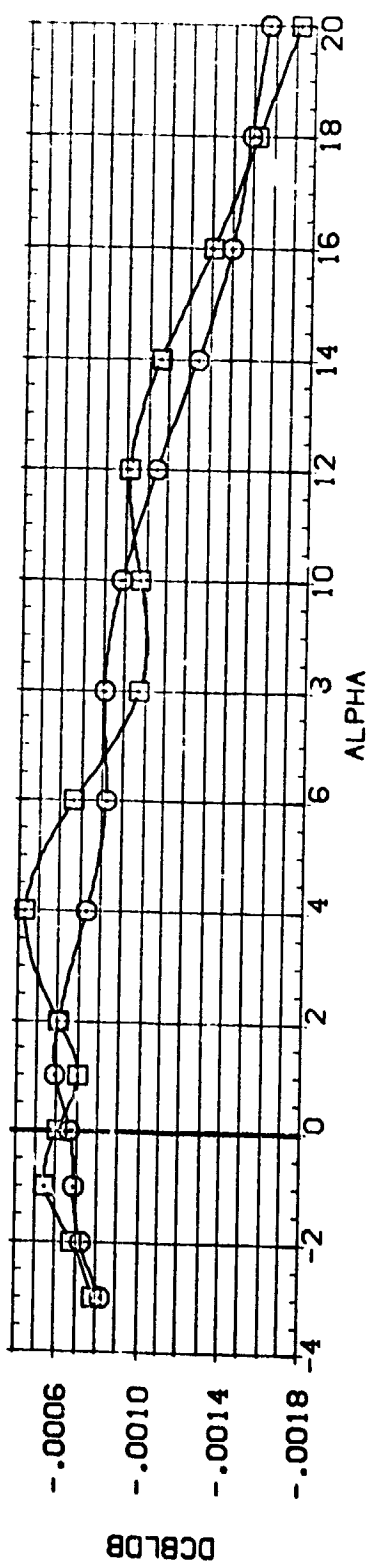
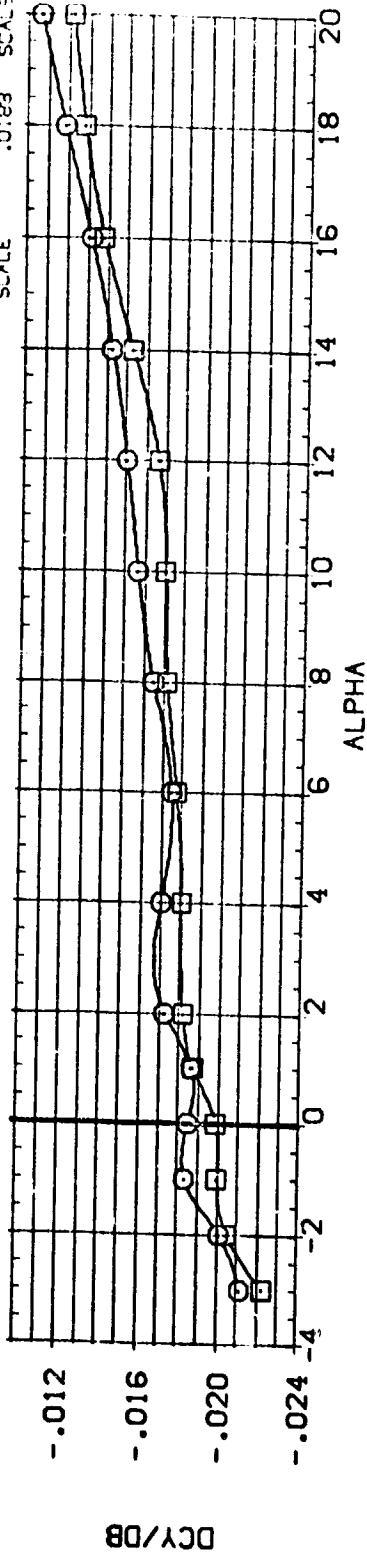
LA-8C: UPVT1040; CRBITTER 0898 V/MOD; NOSE +0'S
LA-8C: UPVT1040; CRBITTER 0898 V/MOD; NOSE +0'S

ELEVTR
-30.000
-30.000

AILERON
.000
.000

RUDDER
10.000
.000

REFERENCE INFORMATION
SREF 136.1808 SQ. IN.
LREF 8.3025 INCHES
BREF 17.5628 INCHES
XMRP 15.7313 INCHES
YMRP .0000 INCHES
ZMRP .0000 INCHES
SCALE .0183



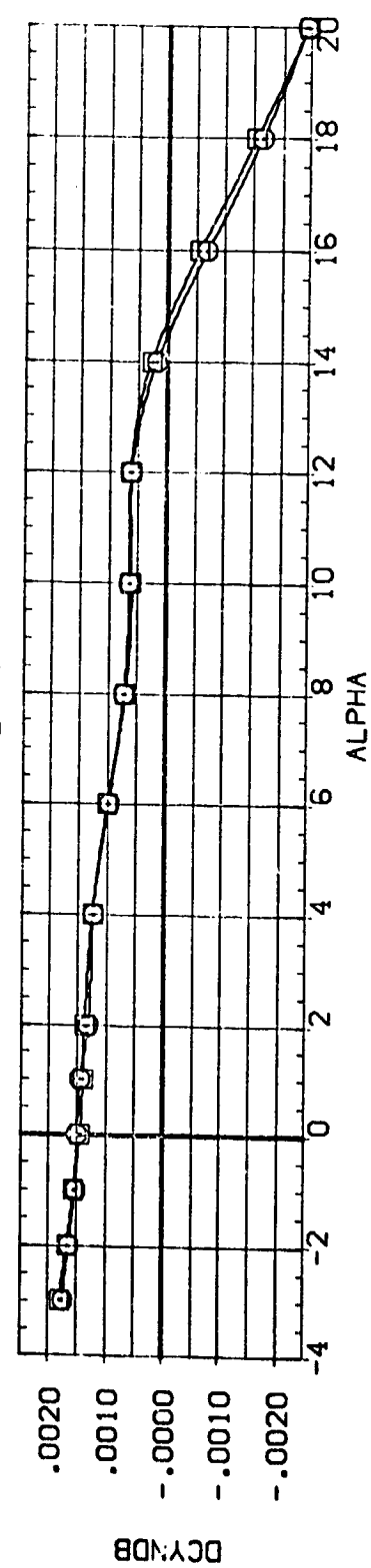
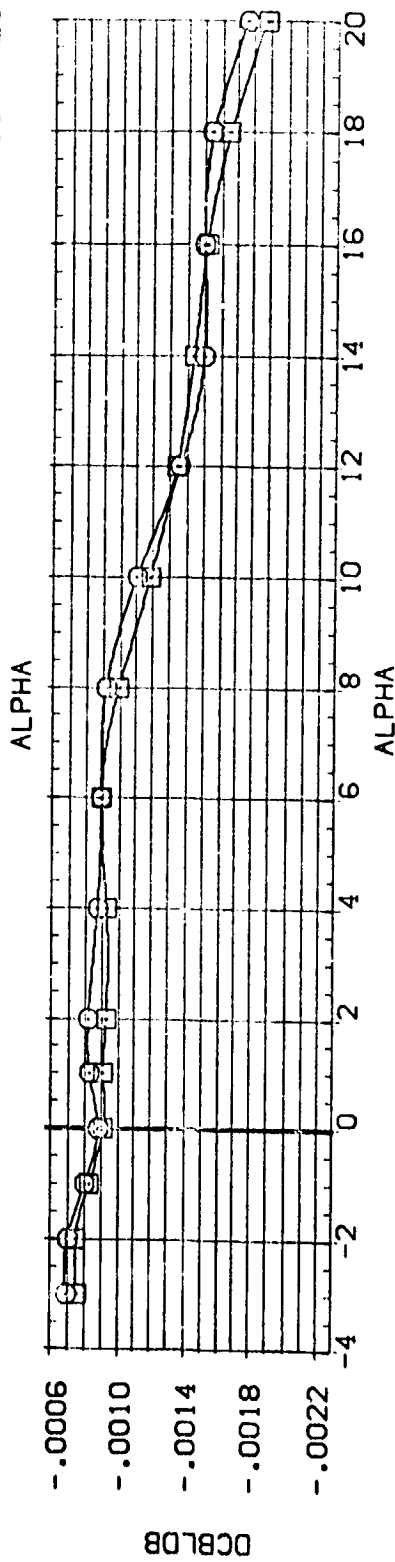
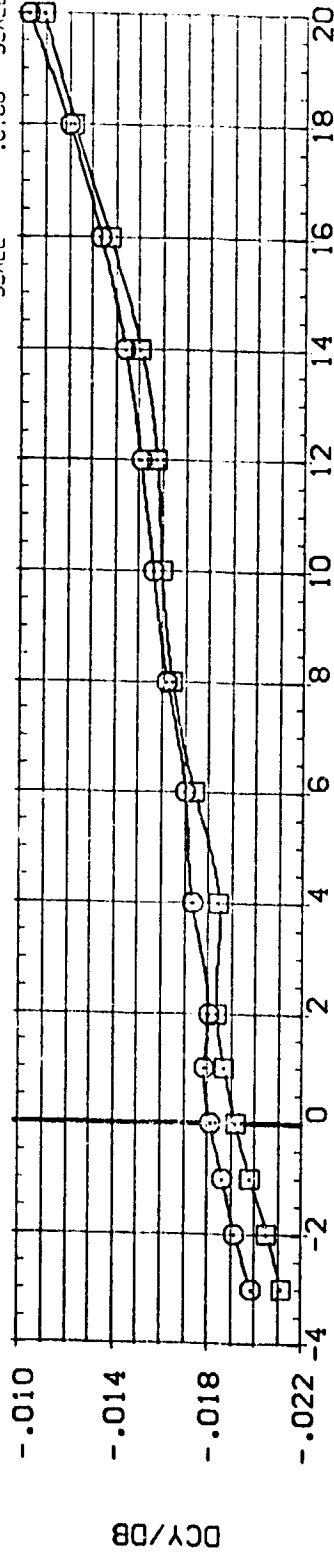
EFFECT OF RUDDER DEFLECTION ON LATERAL-DIRECTIONAL DERIVATIVES

(B)MACH = 2.86

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (MP6C12) □ LA-8C: UPVT1040, ORBITER 0898 V/MCD, NOSE +0MS
 (MP6C04) □ LA-8C: UPVT1040, ORBITER 0898 V/MCD, NOSE +0MS

ELEVTR AIRLON RJODER
 -10.000 10.000 .000
 -10.000 .000

REFERENCE INFORMATION
 SREF 136.18C8 SQ. IN.
 IREF 8.9025 INCHES
 BREF 17.5628 INCHES
 XMRP 15.7313 INCHES
 YMRP .0000 INCHES
 ZMRP .0000 INCHES
 SCALE .0188



EFFECT OF AILERON DEFLECTION ON LATERAL-DIRECTIONAL DERIVATIVES

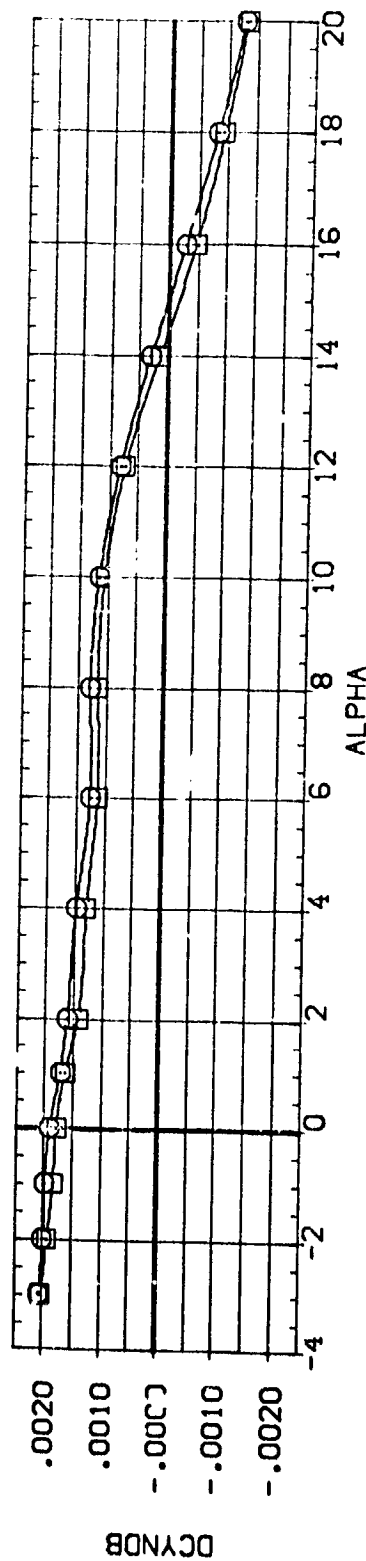
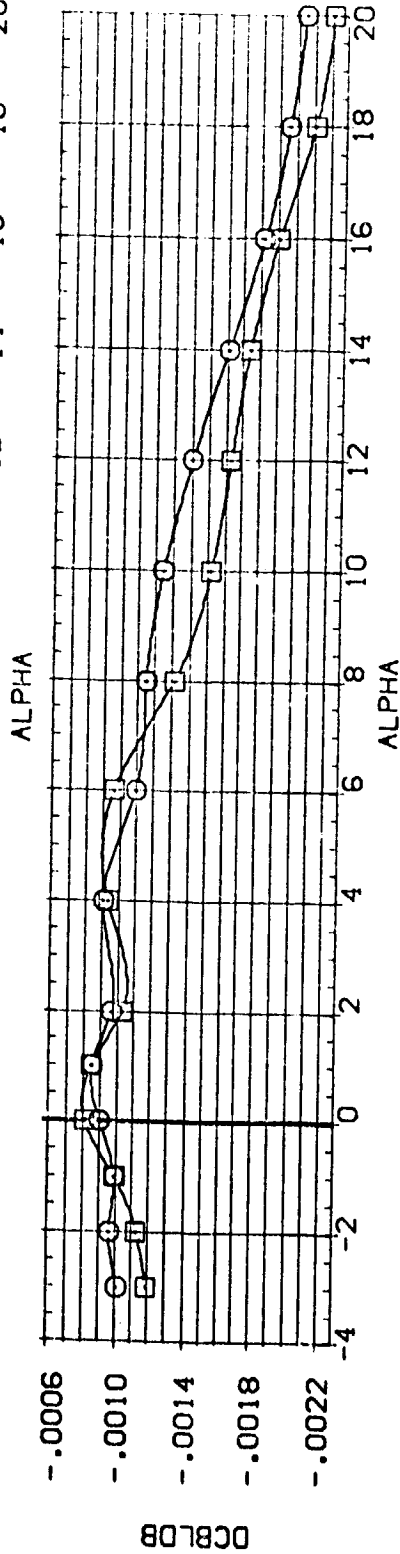
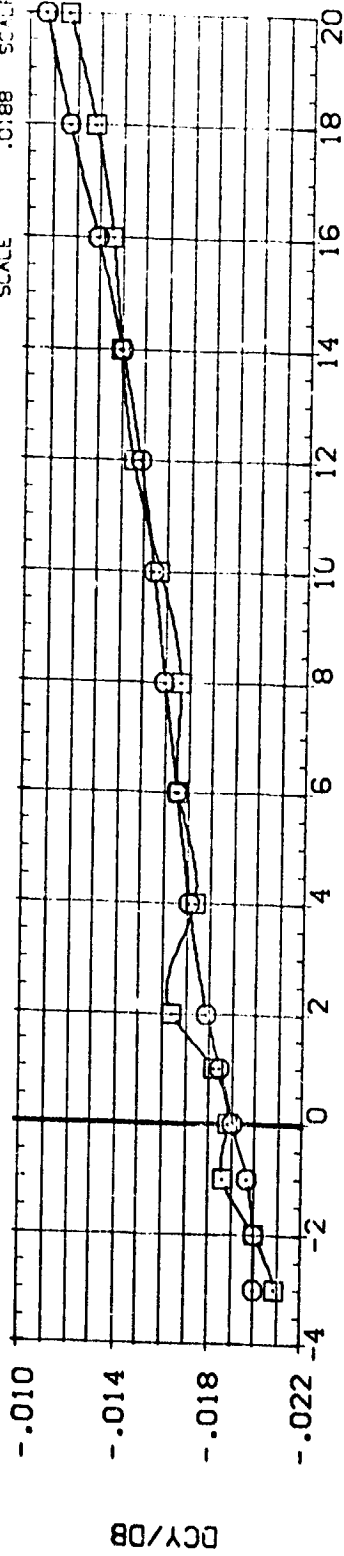
CA1MACH = 1.90

DATA SET SYMBOL
 (NPGC12)
 (NPGC04)

CONFIGURATION DESCRIPTION
 LA-8C: UPVT1040. ORBITER 099B V/MOD. NOSE +0MS
 LA-8C: UPVT1040. ORBITER 099B V/MOD. NOSE +0MS

ELEVTR AILRON RUDDER
 -10.000 10.000 .000
 -10.000 .000 .000

REFERENCE INFORMATION
 SREF 136.1808 SCALIN
 LREF 8.9275 INCHES
 BREF 17.5628 INCHES
 XMRP 15.7313 INCHES
 YMRP .0000 INCHES
 ZMRP .0000 INCHES
 SCALE .0188 SCALE



EFFECT OF AILERON DEFLECTION ON LATERAL-DIRECTIONAL DERIVATIVES

(B)MACH = 2.86

DATA SET SYMBOL
 (AP6213)
 (AP6216)
 (AP6217)
 (AP6218)

CONFIGURATION DESCRIPTION
 LA-8C: UPWT1040, CRG1TER C899 V/MCD, NOSE +CMS
 LA-8C: UPWT1040, CRG1TER C899 V/MCD, NOSE +CMS
 LA-8C: UPWT1040, CRG1TER C899 V/MCD, NOSE +CMS
 LA-8C: UPWT1040, CRG1TER C899 V/MCD, NOSE +CMS
 DATA NOT AVAILABLE
 DATA NOT AVAILABLE

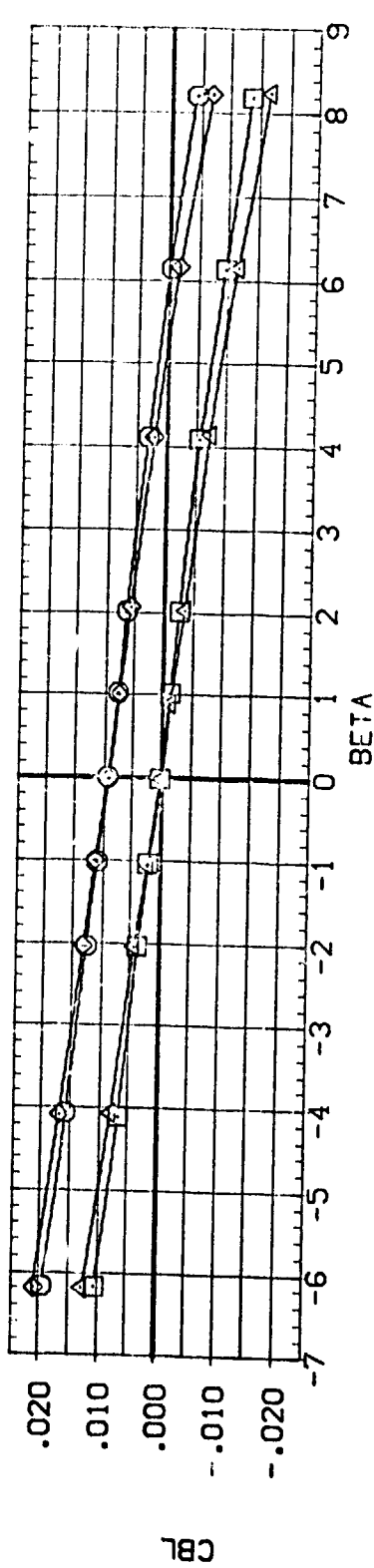
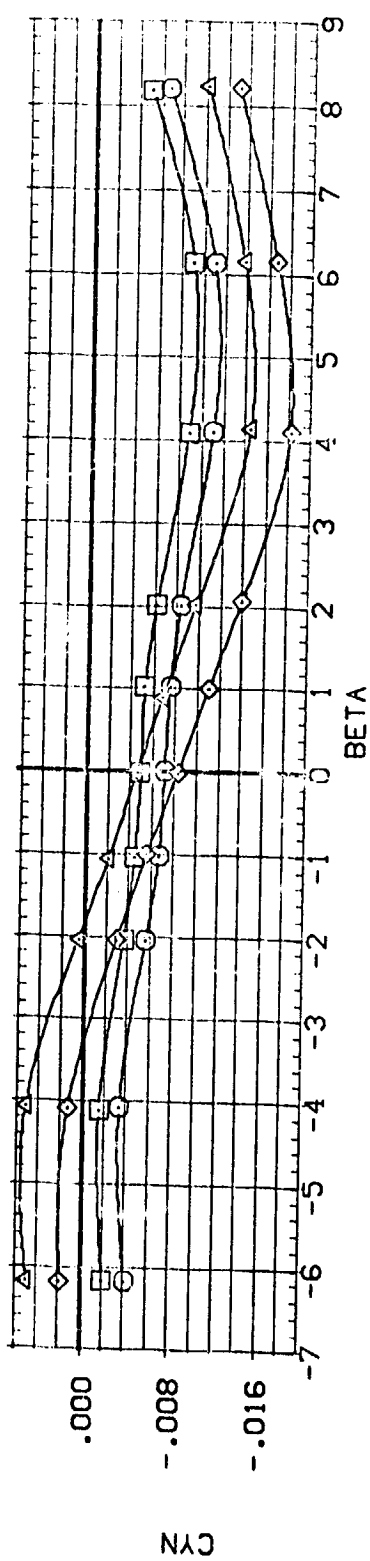
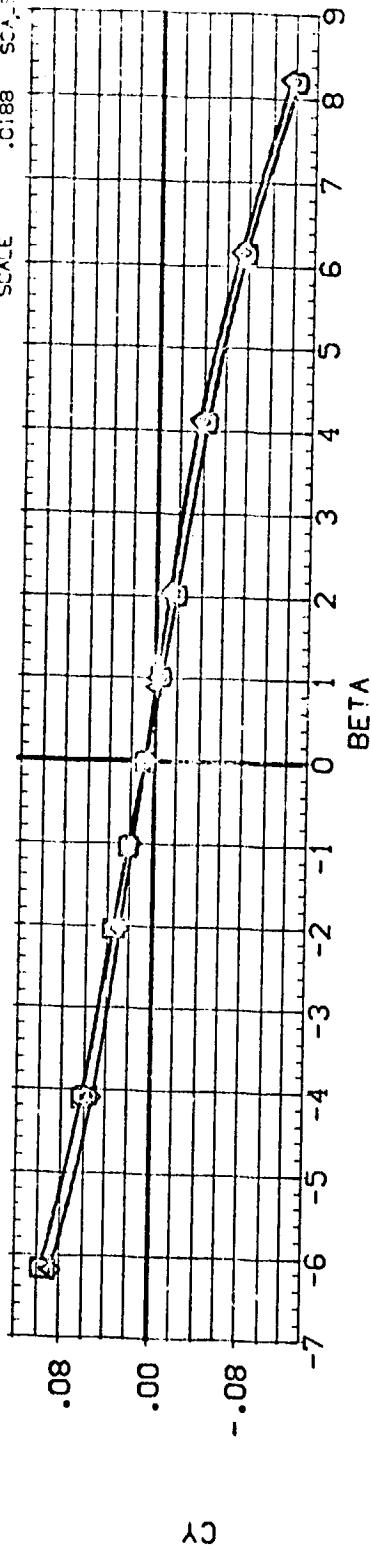
ALPHA 15.000
 15.000
 20.000
 25.000
 25.000
 25.000

ELEVTR -10.000
 -10.000
 -10.000
 -10.000
 -10.000
 -10.000

AILERON 10.000
 .000
 10.000
 10.000
 .000
 .000

RUDDER .000
 .000
 .000
 .000
 .000
 .000

REFERENCE INFORMATION
 SREF 135.1808 SQ. IN.
 LREF 8.9025 INCHES
 BREF 17.5528 INCHES
 XMRP 15.7313 INCHES
 YMRP .0000 INCHES
 ZMRP .0000 INCHES
 SCALE .0188

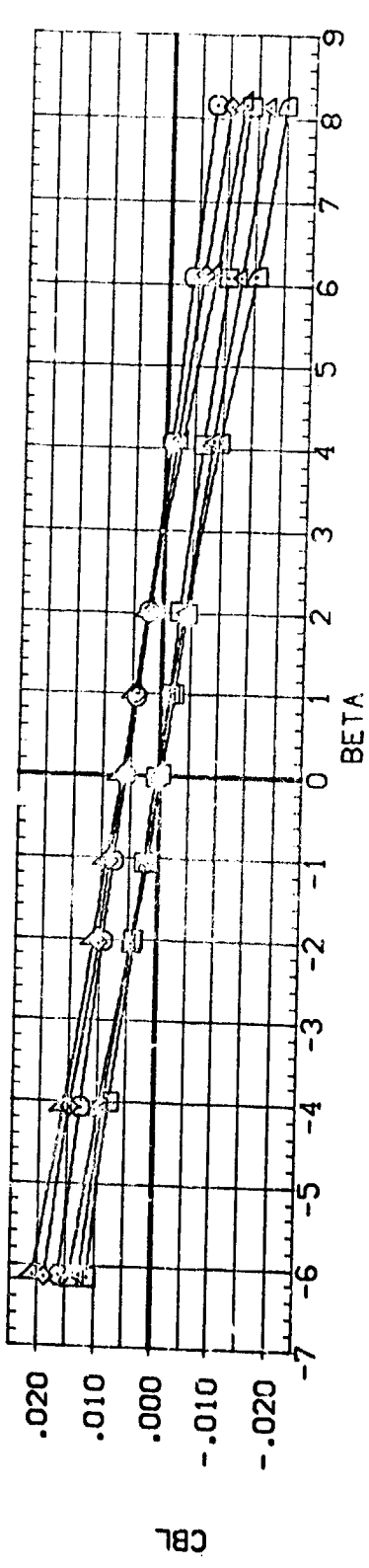
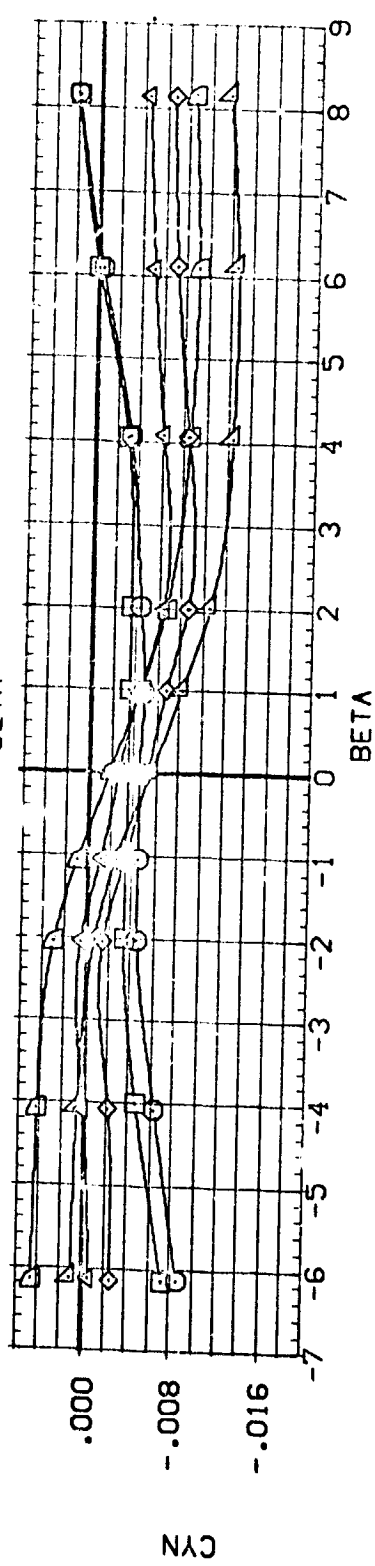
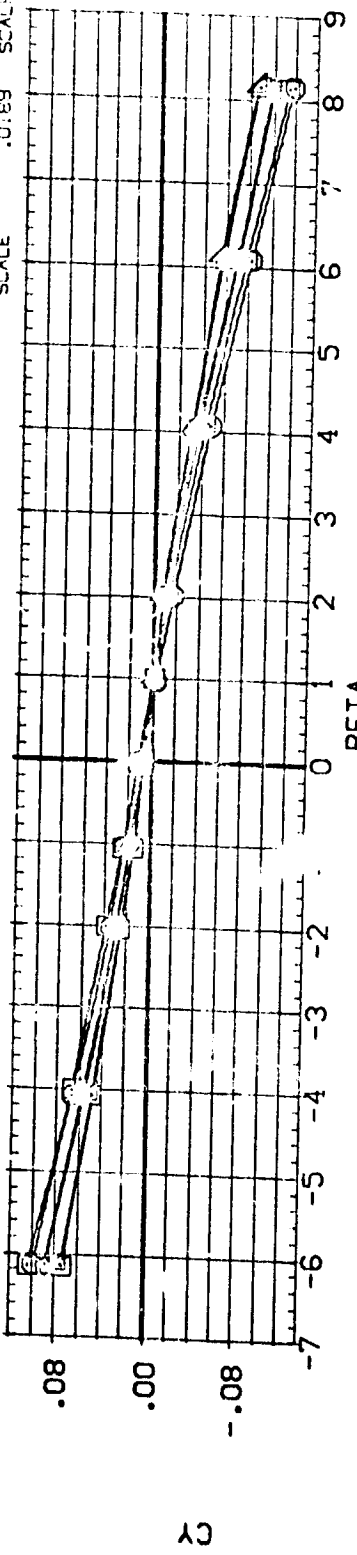


EFFECT OF AILERON DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS IN SIDESLIP

(M)MACH = 1.90

DATA SET SYMBOL CONFIGURATION DESCRIPTION REFERENCE INFORMATION

(AP6213)	LA-8C: UPVT1040, CR3ITER C893 V/MOD, NOSE +CMS	SREF	136.1808	SC.IN.
(AP6216)	LA-8C: UPVT1040, CR3ITER C893 V/MOD, NOSE +CMS	LREF	8.9025	INCHES
(AP6217)	LA-8C: UPVT1040, CR3ITER C893 V/MOD, NOSE +CMS	BREF	17.5628	INCHES
(AP6215)	LA-8C: UPVT1040, CR3ITER C893 V/MOD, NOSE +CMS	YMRP	15.7313	INCHES
(AP6218)	LA-8C: UPVT1040, CR3ITER C893 V/MOD, NOSE +CMS	ZMRP	.0000	INCHES
		SCALE	.0189	SCALE



EFFECT OF AILERON DEFLECTION ON LATERAL-DIRECTIONAL CHARACTERISTICS IN SIDESLIP
 (B)MACH = 2.86 PAGE 28

DATA SET SYMBOL CONFIGURATION DESCRIPTION

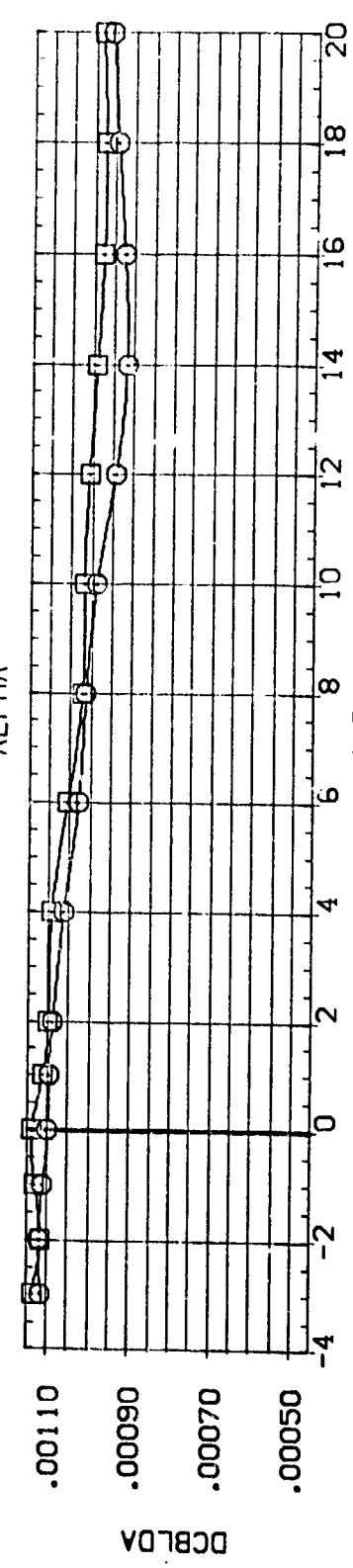
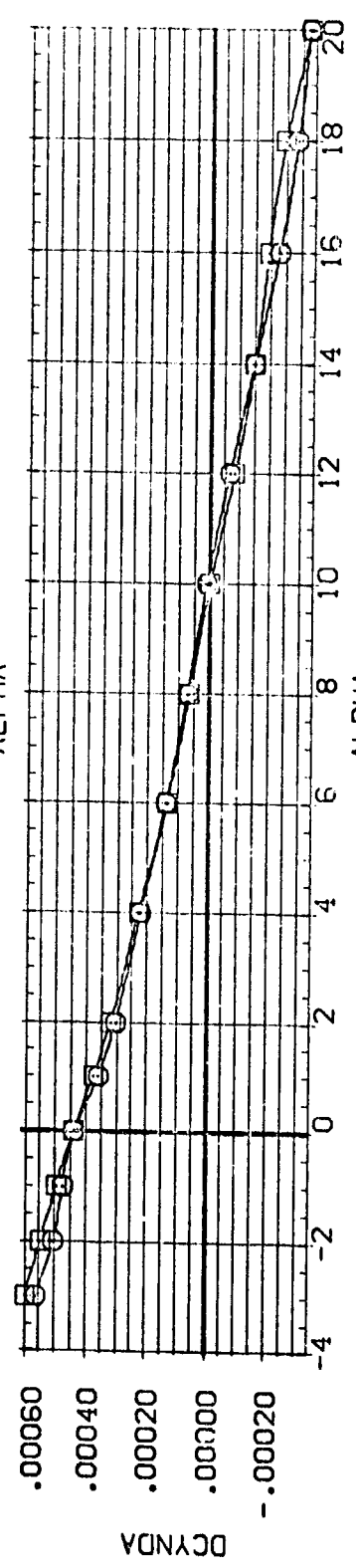
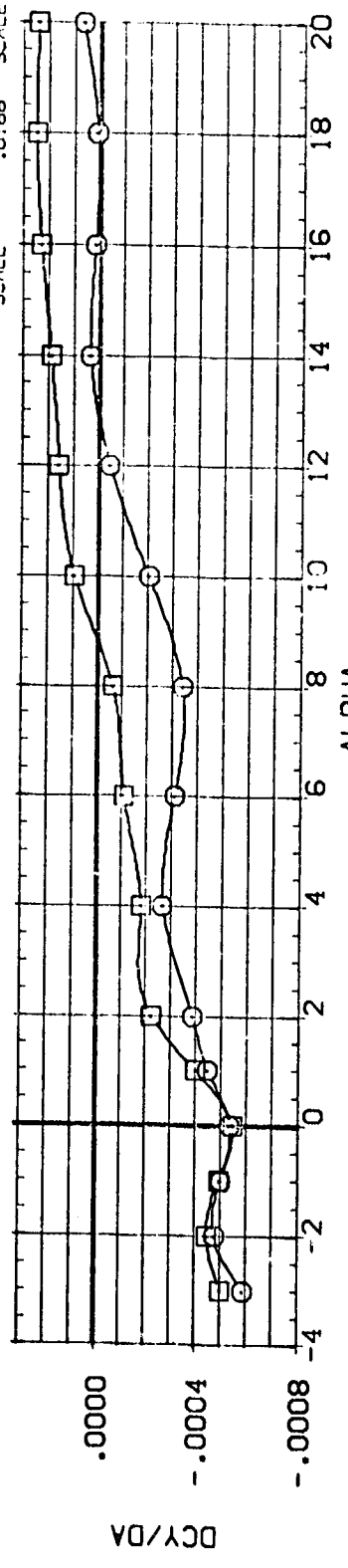
LA-8C: UPVT1040. CRBITER 089J V/MOD. NOSE +0MS
 LA-8C: UPVT1040. CRBITER 089B V/MOD. NOSE +0MS

(CP8C04) □
 (CP8C05) ○

BETA 3.000
 RUOFLR 40.000
 BDFLAP -14.250

REFERENCE INFORMATION

SREF 136.1808 SQ. IN.
 LREF 8.9025 INCHES
 BREF 17.5628 INCHES
 XMRP 15.7313 INCHES
 YMRP .0000 INCHES
 ZMRP .0000 INCHES
 SCALE .0188 INCHES



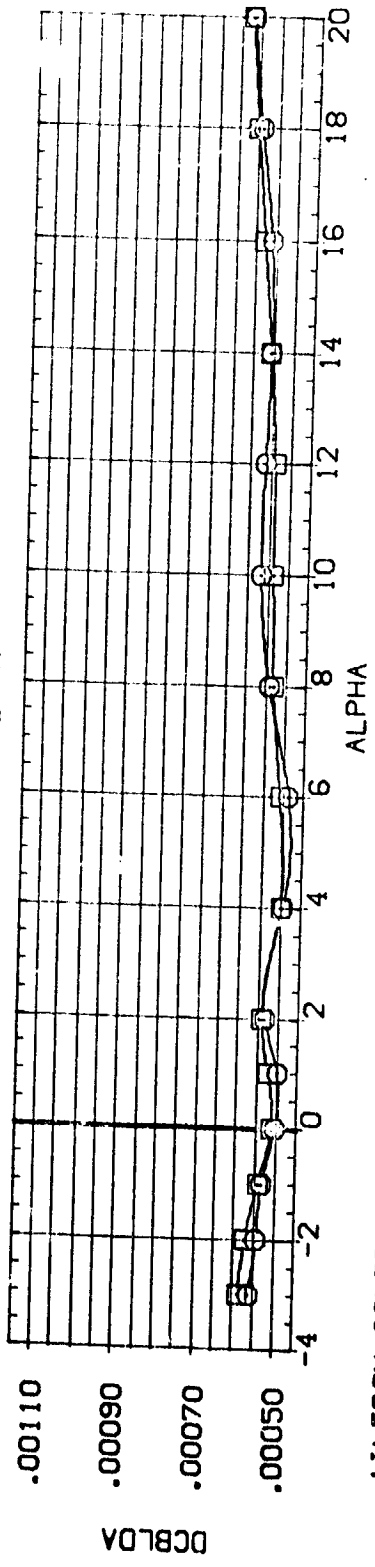
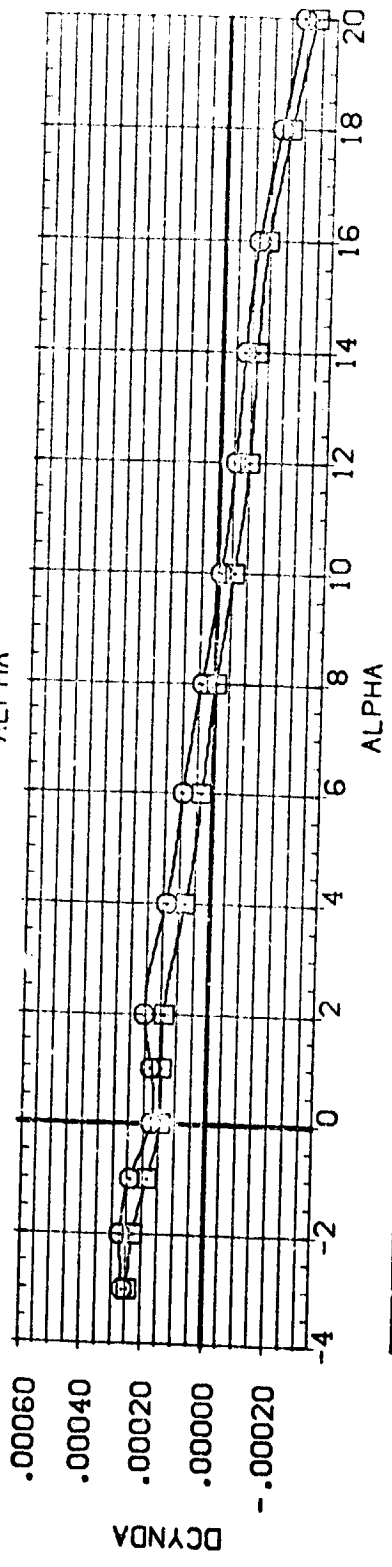
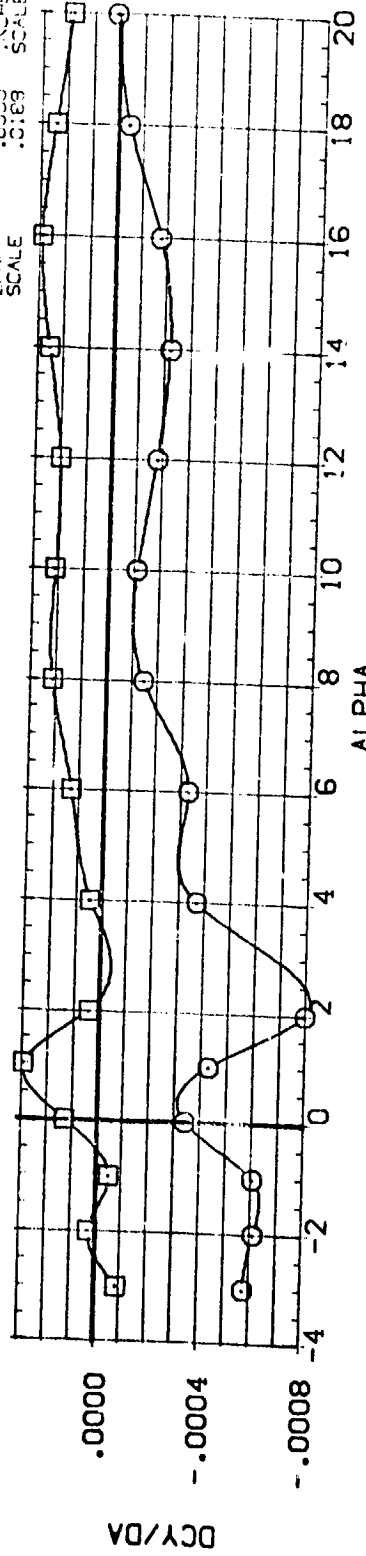
AILERON CONTROL EFFECTIVENESS

(A)MACH = 1.90

DATA SET SYMBOL CONFIGURATION DESCRIPTION
 (CP6C04) □ LA-8C: UPVT1040, CRB1TER 0898 V/MOD, NOSE +CMS
 (CP6C05) □ LA-8C: UPVT1040, CRB1TER 0898 V/MOD, NOSE +CMS

BETA RUJFLR BOFLAP
 3.000 40.000 -14.250
 .000 46.000 -14.250

REFERENCE INFORMATION
 SREF 136.1808 50. IN.
 LREF 8.8025 INCHES
 BREF 17.5628 INCHES
 XMRP .0000 INCHES
 YMRP .0000 INCHES
 ZMRP .0000 INCHES
 SCALE .0169 SCALE



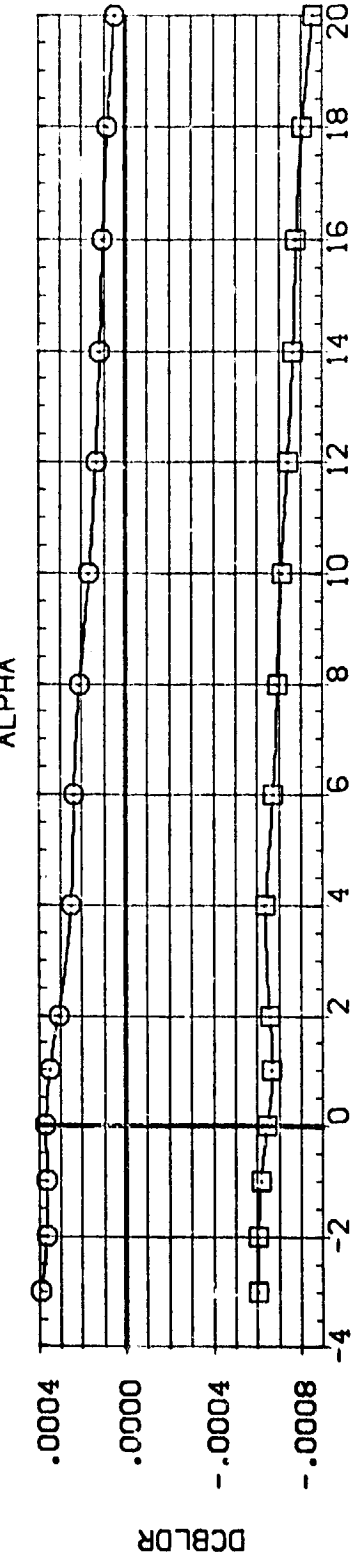
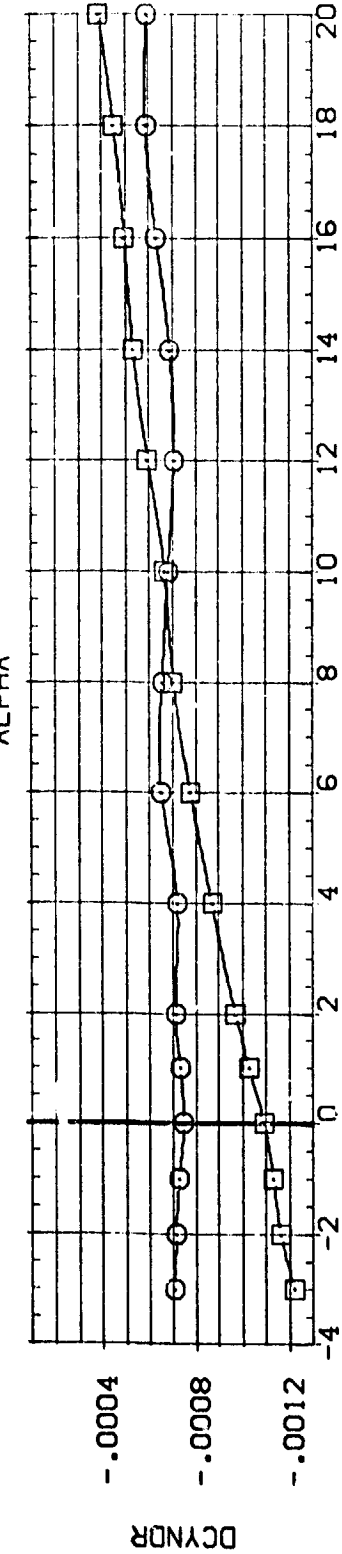
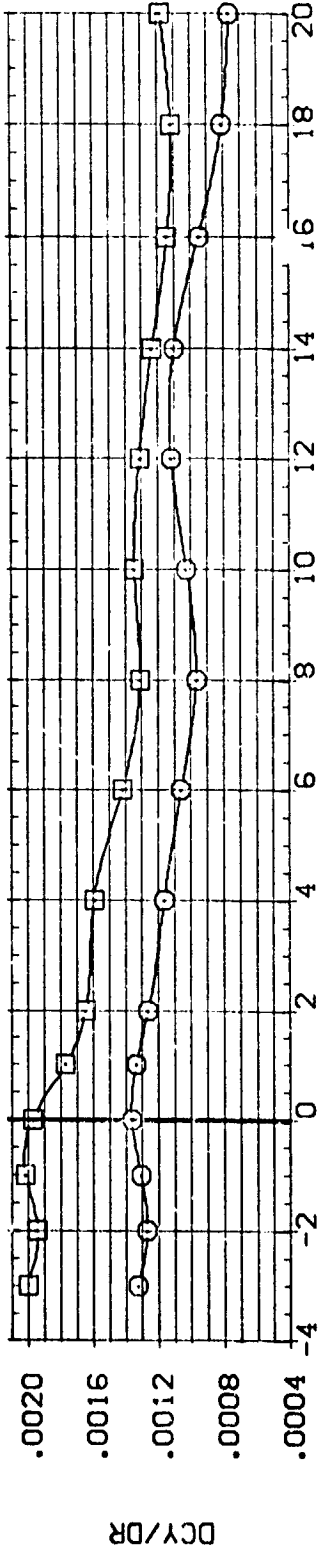
AILERON CONTROL EFFECTIVENESS
 (B)MACH = 2.86

DATA SET SYMBOL (PP6CDB) (PP6C11)

CONFIGURATION DESCRIPTION
 LA-8C. UPVT1040. CRBITER C898 V/MOD. NOSE +0MS
 LA-8C. UPVT1040. CRBITER C893 V/MOD. NOSE +0MS

BETA RUDEL R BDFLAP
 3.000 40.000 40.000 -14.250
 .000 40.000 -14.250

REFERENCE INFORMATION
 SREF 136.1808 SQ. IN.
 LREF 8.9025 INCHES
 BREF 17.5528 INCHES
 XMRP 15.7313 INCHES
 YMRP .0000 INCHES
 ZMRP .0000 INCHES
 SCALE .0188 SCALE



RUDDER CONTROL EFFECTIVENESS

(A)MACH = 1.90

DATA SET SYMBOL: (PP6C08) (PP6C11)

CONFIGURATION DESCRIPTION:
 LA-8C: UPVT1040; CRBITER 0998
 LA-8C: UPVT1040; CRBITER 0998

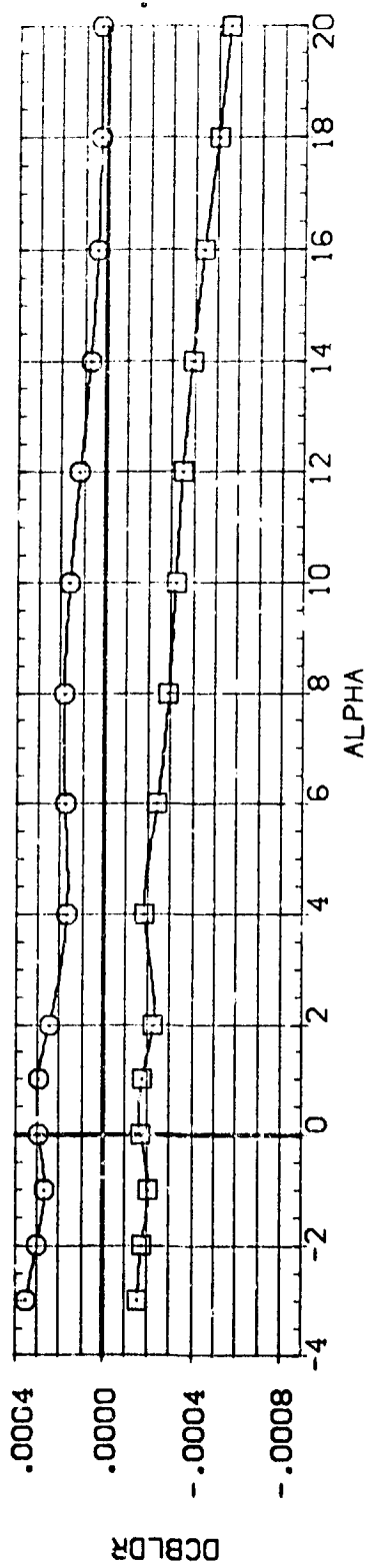
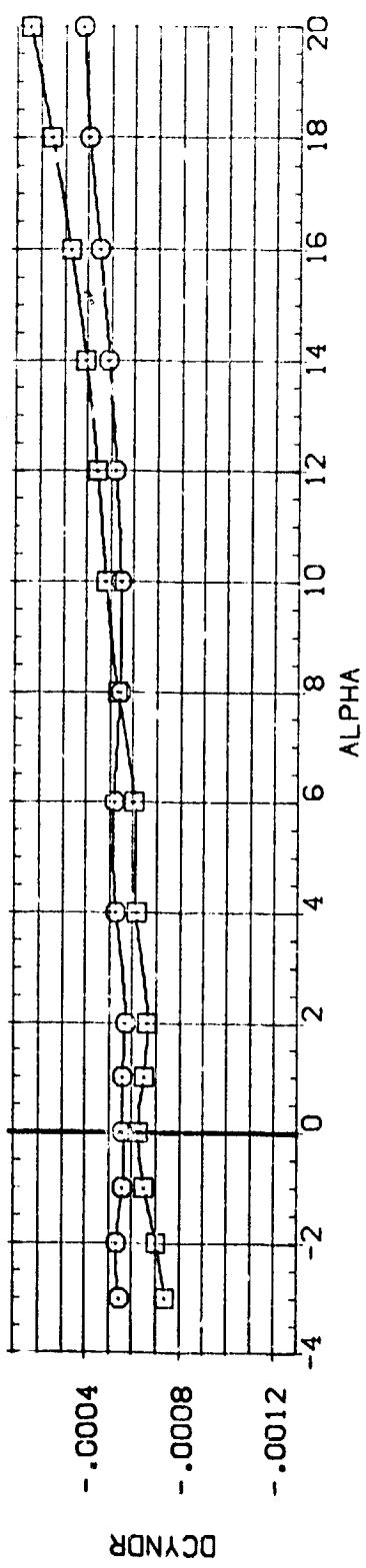
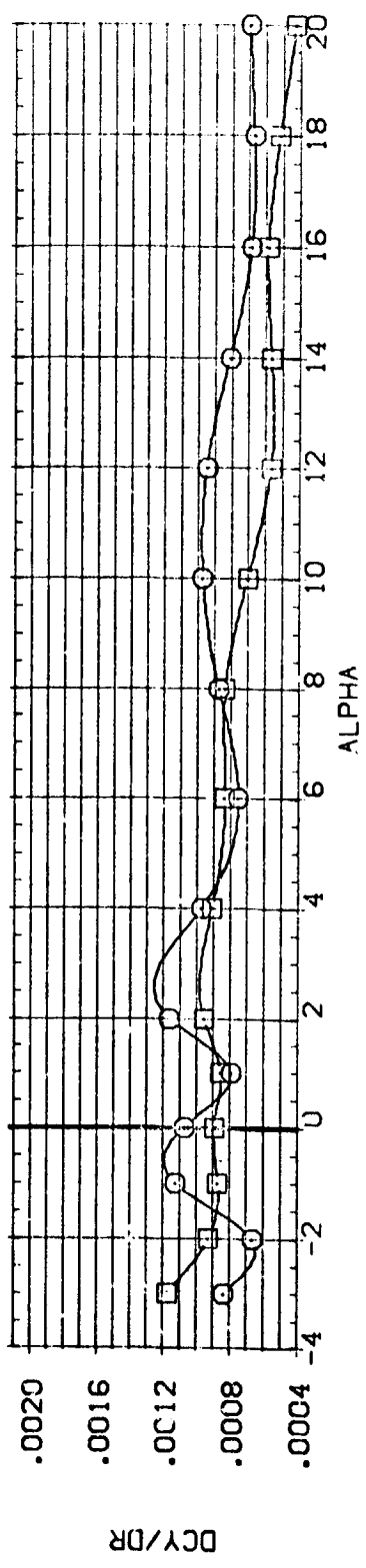
W/MOD: NOSE +0MS
 W/MOD: NOSE +0MS

BETA: 3.000
 .000

RUDFLR: 40.000
 40.000

BDLAP: -14.250
 -14.250

REFERENCE INFORMATION:
 SREF: 136.1808 SQ. IN.
 LREF: 8.9225 INCHES
 XREF: 17.5228 INCHES
 YREF: 15.7313 INCHES
 ZREF: .0000 INCHES
 SCALE: .0188



RUDDER CONTROL EFFECTIVENESS

(B)MACH = 2.86

APPENDIX
TABULATED SOURCE DATA

Plotted data tabulations are
available from DMS on request.

LA-8C, UPWT1040, ORBITER 0898 W/MOD. NOSE +MS

(RP#201) (15 AUG 73)

REFERENCE DATA

SREF = 136.1806 SQ. IN. XMRP = 15.9638 INCHES
 LREF = 8.9725 INCHES YMRP = .0000 INCHES
 BRP = 17.3628 INCHES ZMRP = .0000 INCHES
 SCALE = .0168 SCALE

PARAMETRIC DATA

BETA = .000 ELEVTR = .000
 ATLRON = .000 BDFLAP = -14.250
 RUDGER = .000 RUDFLR = 40.000
 R/L = 5.290 GT-LOC = 2.000

RUN NO. 8/ 0 RN/L = 1.49 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
1.900	-3.239	.00102	-1.14765	.17361	.02101	.00048	-.00451	.01305	-.22292	.26144	-.24432
1.900	-1.423	.00243	-.06716	.17274	.01183	.00035	-.00469	.01243	-.22059	.25843	-.24195
1.900	-3.70	.00048	-.02018	.17219	.00613	.00040	-.00449	.01335	-.22064	.24028	-.24200
1.900	.894	.00083	.02278	.17123	.00077	.00033	-.00436	.01269	-.22079	.24769	-.24213
1.900	1.730	.00119	.06574	.16975	-.00313	.00026	-.00424	.01205	-.22057	.29360	-.24193
1.900	2.797	-.00014	.11069	.16734	-.01137	.00032	-.00421	.01294	-.21791	.36783	-.24192
1.900	4.879	.00003	.19721	.16305	-.02200	.00026	-.00409	.01244	-.21485	.43070	-.23362
1.900	6.987	.00139	.28387	.16034	-.02965	.00010	-.00426	.01187	-.21205	.43406	-.23613
1.900	9.061	.00215	.36624	.15787	-.03629	-.00005	-.00428	.01133	-.20925	.43476	-.24130
1.900	13.242	.00301	.53138	.15298	-.04316	.00008	-.00450	.01103	-.22248	.32307	-.25455
1.900	17.454	.00385	.70857	.14505	-.05049	-.00027	-.00463	.01076	-.23044	.46146	-.25986
1.900	20.526	.00371	.84026	.13809	-.05626	-.00037	-.00441	.01040	-.22517	.65285	-.25723
GRADIENT	-.00020	.04232	-.00130	-.00535	-.00002	-.00007	-.00007	-.00006	.00088	.02088	.00100

RUN NO. 1/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
2.860	-4.368	-.01828	-.15468	.13912	.03481	.00186	-.00486	.01474	-.09022	-.15184	-.05283
2.860	-2.565	-.01563	-.10652	.13678	.03389	.00195	-.00514	.01295	-.09335	-.15188	-.04267
2.860	-1.548	-.01697	-.07545	.13570	.03320	.00162	-.00480	.01311	-.09336	-.15188	-.03307
2.860	-.499	-.01775	-.04188	.13430	.03274	.00170	-.00463	.01325	-.09336	-.15188	-.02987
2.860	.531	-.01617	-.01569	.13280	.03176	.00131	-.00454	.01146	-.09361	-.15189	-.02996
2.860	1.568	-.01859	.01309	.13206	.03103	.00139	-.00448	.01345	-.09359	-.15510	-.02993
2.860	3.620	-.01946	.07276	.12878	.02806	.00141	-.00432	.01366	-.09682	-.15511	-.03956
2.860	5.680	-.01579	.13719	.12635	.02718	.00103	-.00430	.01012	-.09683	-.15511	-.05559
2.860	7.737	-.01616	.19848	.12407	.02436	.00125	-.00402	.00943	-.09663	-.15511	-.06518
2.860	11.676	-.01673	.33547	.11911	.02161	.00028	-.00393	.00986	-.10322	-.15511	-.08117
2.860	16.011	-.01982	.48472	.11475	.01728	.00002	-.00312	.01046	-.10643	-.15632	-.10388
2.860	20.149	-.02123	.63793	.10400	.00992	-.00002	-.00214	.00918	-.10644	-.16153	-.10360
2.860	24.309	-.02038	.80867	.10082	.00405	-.00005	-.00152	.00698	-.10323	-.15511	-.12277
2.860	28.480	-.02183	.98175	.09322	-.00159	-.00076	-.00076	.00667	-.09683	-.14669	-.12277
GRADIENT	-.00024	.02899	-.00127	-.00081	-.00008	-.00008	-.00009	-.00011	-.00062	-.00074	.00005

LA-9C, UPWT1040, ORBITER 0898 W/MOD. NOSE NOMS

(RP6202) (15 AUG 73)

REFERENCE DATA

SREF = 136.1608 SQ. IN. XMRP = 15.9638 INCHES
LREF = 6.9025 INCHES YMRP = .0000 INCHES
BREF = 17.5628 INCHES ZMRP = .0000 INCHES
SCALE = .0188 SCALE

PARAMETRIC DATA

BETA = 3.000 ELEVTR = .000
AILRON = .000 BDFLAP = -14.250
RUDDER = .000 RUOFLR = 40.000
K/L = 5.290 GT-LOC = 2.000

RUN NO. 9/ 0 RN/L = 1.49 GRADIENT INTERVAL = -5.00/ 5.00

Table with columns: MACH, ALPHA, BETA, CN, CA, CLM, CBL, CYN, CY, CFB1, CPB2, CPC. Rows for MACH 1.900 to 2.900.

RUN NO. 2/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

Table with columns: MACH, ALPHA, BETA, CN, CA, CLM, CBL, CYN, CY, CFB1, CPB2, CPC. Rows for MACH 2.000 to 2.900.

REFERENCE DATA
 SREF = 136.1606 SQ.IN. XMRP = 15.9638 INCHES BETA = .000 ELEVTR = -10.000
 LREF = 8.9025 INCHES YMRP = .0000 INCHES AIRRON = .000 BDFLAP = -14.250
 BREF = 1.3628 INCHES ZMRP = .0000 INCHES RUDDER = .000 RUDFLR = 40.000
 SCALE = .0166 SCALE K/L = 5.290 GT-LOC = 2.000

PARAMETRIC DATA

BETA = .000 ELEVTR = -10.000
 AIRRON = .000 BDFLAP = -14.250
 RUDDER = .000 RUDFLR = 40.000
 K/L = 5.290 GT-LOC = 2.000

RUN NO. 10/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
1.900	-3.263	.00170	-1.9024	1.7600	.04443	.00096	-.00516	.01446	-.19005	-.27997	-.24041
1.900	-1.382	.00243	-1.10717	1.7410	.03474	.00061	-.00498	.01327	-.18574	-.27637	-.23384
1.900	-1.385	.00273	-.06187	1.7279	.03039	.00080	-.00486	.01267	-.17992	-.27610	-.22813
1.900	.696	.00246	-.01650	1.7259	.02444	.00078	-.00458	.01203	-.16943	-.27619	-.22559
1.900	1.715	.00404	.02688	1.7210	.01916	.00070	-.00427	.01133	-.16410	-.27619	-.22293
1.900	2.778	.00206	-.07350	1.6957	.01381	.00082	-.00453	.01213	-.16155	-.27230	-.22232
1.900	4.678	.00284	1.5840	1.6373	.00324	.00017	-.00455	.01130	-.16287	-.27037	-.22166
1.900	6.974	.00301	2.2613	1.6044	-.00623	.00060	-.00448	.01108	-.14992	-.27705	-.21639
1.900	9.047	.00376	3.3141	1.5695	-.01393	.00044	-.00451	.01052	-.12652	-.27458	-.21915
1.900	13.252	.00638	4.9599	1.4871	-.02024	.00034	-.00493	.00941	-.10795	-.26924	-.22438
1.900	17.440	.00597	6.6855	1.3	-.02506	.00044	-.00485	.00922	-.10984	-.26690	-.23188
1.900	20.920	.00518	8.0124	1.3369	-.02922	.00027	-.00458	.00889	-.14199	-.26909	-.23741
GRADIENT		.00012	.04293	-.00138	-.00507	-.00001	.00006	-.00036	.000402	.00108	.00248

RUN NO. 14/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
2.600	-4.352	-.01893	-1.6479	1.4013	.04658	.00240	-.00493	.01365	.05525	-.15311	-.12080
2.600	-2.538	-.02030	-1.1222	1.3683	.04438	.00233	-.00494	.01675	.00723	-.15312	-.12083
2.600	-1.535	-.01760	-.08616	1.3536	.04415	.00209	-.00492	.01406	-.01165	-.15308	-.12394
2.600	-.491	-.01661	-.06244	1.3386	.04280	.00161	-.00466	.01229	-.01808	-.15309	-.12395
2.600	.527	-.01904	-.02655	1.3223	.04183	.00146	-.00460	.01429	-.02133	-.15310	-.12397
2.600	1.552	-.01864	-.00028	1.3088	.04089	.00154	-.00447	.01346	-.02131	-.15309	-.12396
2.600	3.610	-.01946	.05442	1.2750	.03747	.00124	-.00431	.01364	-.01504	-.15311	-.12080
2.600	5.674	-.02038	1.2385	1.2425	.03768	.00081	-.00412	.01391	-.00856	-.15310	-.11757
2.600	7.738	-.02073	1.8236	1.2173	.03483	.00095	-.00381	.01319	-.01032	-.15313	-.11767
2.600	11.866	-.01908	3.1470	1.1568	.03257	.00092	-.00396	.01175	.05512	-.15312	-.10805
2.600	15.989	-.02436	4.4654	1.1116	.03085	.00059	-.00311	.01110	.01110	-.15947	-.13025
2.600	20.143	-.02330	6.0516	1.0340	.02563	.00056	-.00269	.01190	-.02098	-.15948	-.13347
2.600	24.256	-.02282	7.6657	.09524	.02401	-.00020	-.00193	.00976	-.07276	-.15305	-.13988
2.600	28.466	-.02338	9.4244	.08808	.01945	-.00011	-.00121	.00855	-.04664	-.14662	-.13348
GRADIENT		-.00301	.02756	-.00136	-.00109	-.00016	.00009	-.00034	-.00811	.00000	-.00014

REFERENCE DATA

SREF = 136.1608 50. IN. XMRP = 15.9638 INCHES
 LREF = 6.9025 INCHES YMRP = .0000 INCHES
 BREF = 17.3628 INCHES ZMRP = .0000 INCHES
 SCALE = .0108 SCALE

PARAMETRIC DATA

BETA = 3.000 ELEVTR = -10.000
 AIRLON = .000 BDFLAP = -14.250
 RUDDER = .000 RUDFLR = 40.000
 K/L = 5.290 GT-LOC = 2.000

RUN NO. 11/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
1.900	-3.247	3.03709	-1.8264	1.7525	.04223	-.00136	-.00049	-.04979	-.15028	-.28253	-.24592
1.900	-1.404	3.03361	-1.0234	1.7365	.03392	-.00176	-.00097	-.04742	-.13138	-.28235	-.24268
1.900	-3.71	3.03456	-0.9752	1.7282	.02879	-.00191	-.00117	-.04584	-.12316	-.28223	-.24252
1.900	.696	3.03326	-.01053	1.7158	.02237	-.00199	-.00110	-.04495	-.11522	-.28223	-.24252
1.900	1.716	3.03318	.03067	1.7054	.01766	-.00208	-.00133	-.04411	-.11261	-.28224	-.24254
1.900	2.787	3.03300	.07364	1.6901	.01118	-.00214	-.00178	-.04398	-.11157	-.28229	-.24260
1.900	4.683	3.03293	.16393	1.6447	.00160	-.00200	-.00176	-.04306	-.11242	-.27703	-.24002
1.900	6.969	3.03289	.24997	1.6065	-.001817	-.00218	-.00263	-.03986	-.12891	-.27446	-.24023
1.900	9.045	3.03432	.33372	1.5671	-.01394	-.00287	-.00332	-.03820	-.13166	-.27186	-.24020
1.900	13.260	3.03725	.49638	1.4724	-.01960	-.00362	-.00415	-.03714	-.13305	-.26385	-.23749
1.900	17.436	3.04631	.67380	1.3697	-.02611	-.00451	-.00927	-.02894	-.11296	-.25585	-.23215
1.900	20.324	3.05336	.80024	1.2900	-.02950	-.00559	-.01309	-.02301	-.12874	-.25315	-.22945
GRADIENT		-.00045	.04258	-.00126	-.00510	-.00008	-.00014	.00082	-.00342	.00052	.00028

RUN NO. 15/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
2.800	-4.336	3.05981	-.16747	1.3922	.04368	-.00092	.00003	-.04918	.00442	-.14986	-.12711
2.800	-2.572	3.05659	-.11740	1.3610	.04127	-.00122	.00024	-.04534	-.00508	-.14985	-.12368
2.800	-1.590	3.05626	-.07337	1.3462	.04081	-.00113	-.00006	-.04436	-.01794	-.14986	-.12390
2.800	-.508	3.05493	-.03261	1.3292	.04012	-.00113	.00001	-.04325	-.02443	-.14987	-.12393
2.800	.515	3.05541	-.02629	1.3164	.03916	-.00069	-.00015	-.04322	-.02114	-.14986	-.12393
2.800	1.542	3.05096	-.00473	1.3019	.03864	-.00140	-.00063	-.03761	-.02114	-.14986	-.12711
2.800	3.606	3.05309	.05485	1.2774	.03564	-.00168	-.00097	-.03842	-.01801	-.14987	-.13033
2.800	5.664	3.05240	.11696	1.2478	.03370	-.00210	-.00135	-.03644	-.01160	-.15308	-.13032
2.800	7.730	3.05267	.18314	1.2145	.03379	-.00286	-.00101	-.03715	-.01475	-.15307	-.13031
2.800	11.848	3.05257	.30824	1.1580	.03083	-.00399	-.00227	-.03232	-.02441	-.15308	-.12713
2.800	16.010	3.05663	.46538	1.0962	.02886	-.00516	-.00509	-.02672	-.01476	-.15950	-.12391
2.800	20.140	3.06057	.60972	1.0299	.02412	-.00613	-.00725	-.02293	-.01479	-.15950	-.13032
2.800	24.268	3.05953	.76461	.05489	.02373	-.00622	-.00853	-.01794	-.03080	-.15315	-.13352
2.800	28.459	3.06412	.94222	.08796	.01931	-.00839	-.00832	-.01940	-.03084	-.14666	-.13354
GRADIENT		-.00090	.02804	-.00144	-.00093	-.00008	-.00018	.00141	-.00232	-.00000	-.00048

LA-8C, UPWT1040, ORBITER 0898 W/MOD. NOSE +ONS

(RP&ZUS) (15 AUG 73)

REFERENCE DATA

SREF = 136.1008 SQ.IN. XMRP = 15.9636 INCHES
 LREF = 8.9023 INCHES YMRP = .0000 INCHES
 BREF = 17.5828 INCHES ZMRP = .0000 INCHES
 SCALE = .0188 SCALE

PARAMETRIC DATA

BETA = .000 ELEVTR = -20.000
 AILRON = .000 BDFLAP = -14.250
 RUDDER = .000 RUOFLR = 40.000
 K/L = 5.290 GT-LOC = 2.000

RUN NO. 26/ 0 RN/L = 1.49 GRADIENT INTERVAL = -.5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
1.900	-3.255	.00852	-2.3561	.18768	.06928	.00044	-.00540	.00981	-.14618	-.27860	-.25476
1.900	-1.416	.00971	-1.15674	.18577	.05937	.00048	-.00530	.00932	-.13377	-.20044	-.25376
1.900	-3.77	.00728	-.10741	.18457	.05453	.00033	-.00498	.00951	-.12304	-.26040	-.25105
1.900	.670	.00823	-.06180	.18293	.04826	.00037	-.00499	.00879	-.11111	-.28096	-.25175
1.900	1.722	.00742	-.01658	.18106	.04297	.00029	-.00481	.00890	-.10679	-.28135	-.24960
1.900	2.786	.00775	.02840	.17855	.03785	.00035	-.00468	.00825	-.09403	-.28152	-.24717
1.900	4.848	.00792	.11647	.17272	.02632	.00034	-.00457	.00773	-.07478	-.27327	-.24151
1.900	6.966	.00691	.20158	.16845	.01939	.00040	-.00444	.00802	-.05775	-.27817	-.24363
1.900	9.046	.00943	.28822	.16382	.01175	-.00004	-.00462	.00668	-.03161	-.27768	-.24326
1.900	13.237	.01062	.45548	.15308	.00901	-.00010	-.00468	.00562	-.00925	-.28327	-.24066
1.900	17.445	.01108	.63106	.14434	.00104	-.00025	-.00498	.00612	-.05463	-.28867	-.25670
1.900	20.521	.01081	.76164	.13517	-.00195	-.00034	-.00454	.00504	-.07662	-.28626	-.25967
GRADIENT		-.00010	.04357	-.00182	-.00528	-.00002	.00011	-.00026	.00890	.00044	.00158

RUN NO. 30/ 0 RN/L = 1.50 GRADIENT INTERVAL = -.5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CFC
2.660	-4.376	.00619	-.19136	.14800	.06622	.00163	-.00505	.01005	.00579	-.15486	-.12563
2.660	-2.584	.00650	-.14004	.14406	.06241	.00155	-.00494	.00931	-.00331	-.15802	-.12549
2.660	-1.529	.00574	-.10666	.14174	.06114	.00155	-.00477	.00946	-.00308	-.15800	-.12221
2.660	-.497	.00445	-.07989	.13982	.05994	.00165	-.00472	.01046	-.00374	-.15806	-.11921
2.660	.514	.00720	-.05619	.13827	.05784	.00158	-.00467	.00771	-.00398	-.16129	-.11929
2.660	1.561	.00529	-.02036	.13630	.05688	.00131	-.00446	.00879	-.00083	-.16130	-.11611
2.660	3.652	.00443	.04159	.13169	.05414	.00101	-.00430	.00902	.01537	-.16449	-.10964
2.660	5.680	.00296	.10113	.12745	.05032	.00099	-.00395	.00927	.01550	-.16127	-.10960
2.660	7.740	.00437	.16236	.12419	.04831	.00064	-.00383	.00760	.04108	-.15806	-.10960
2.660	11.865	.00263	.29000	.11758	.04731	.00047	-.00380	.00896	.06012	-.15807	-.11284
2.660	16.010	.00299	.43491	.11199	.04655	.00016	-.00345	.00763	.02173	-.16128	-.13203
2.660	20.157	.00039	.58563	.10366	.04609	.00004	-.00249	.00732	-.01042	-.16130	-.13847
2.660	24.303	.00123	.74499	.09501	.04518	-.00020	-.00192	.00512	-.06140	-.15808	-.14164
2.660	28.475	.00083	.91151	.08721	.04572	-.00059	-.00078	.00301	-.05497	-.14845	-.13523
GRADIENT		-.00019	.02888	-.00199	-.00147	-.00007	.00010	-.00016	.00106	-.00114	.00203

LA-6C, UPWT1040, ORBITER 0898 W/MOD. NOSE +ONS

(RP6206) (15 AUG 73)

REFERENCE DATA

SREF = 136.1848 SQ.IN. XMRP = 15.9638 INCHES
 LREF = 9.9025 INCHES YMRP = .0000 INCHES
 BREF = 17.5628 INCHES ZMRP = .0000 INCHES
 SCALE = .0188 SCALE

PARAMETRIC DATA

BETA = 3.000 ELEVTR = -20.000
 AIRLON = .000 BDFLAP = -14.250
 RUDDER = .000 RUOFLR = 40.000
 K/L = 5.290 GT-LOC = 2.000

RUN NO. 29/ 0 RV/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
1.900	-3.255	3.040/9	-23016	.06988	-.00165	.00008	-.03445	-.11126	-.27837	-.25448
1.900	-1.421	3.03891	-15181	.18483	-.00183	-.00022	-.05200	-.10856	-.28100	-.25445
1.900	-.390	3.03878	-10460	.05558	-.00190	-.00046	-.05115	-.10856	-.28366	-.25445
1.900	.660	3.03811	-.05968	.18185	-.00204	-.00054	-.05029	-.11388	-.28633	-.25446
1.900	1.724	3.03812	-.01461	.18032	-.00227	-.00104	-.04871	-.11921	-.28899	-.25713
1.900	2.772	3.03789	.03257	.17825	-.00202	-.00100	-.04856	-.11144	-.28916	-.25721
1.900	4.862	3.03778	.11889	.02781	-.00206	-.00174	-.04613	-.10627	-.28645	-.25198
1.900	6.969	3.03764	.20908	.01840	-.00209	-.00248	-.04369	-.11427	-.28382	-.25466
1.900	9.052	3.03804	.29305	.01265	-.00255	-.00504	-.04202	-.12220	-.28382	-.25466
1.900	13.246	3.04107	.45968	.00660	-.00406	-.00403	-.04022	-.09828	-.27846	-.25195
1.900	17.453	3.04015	.63493	.00010	-.00468	-.00918	-.03201	-.11156	-.27849	-.24933
1.900	20.523	3.03717	.76345	.13182	-.00186	-.00506	-.02751	-.13005	-.27317	-.24667
GRADIENT	-.00034	.04317	-.00168	-.00526	-.00006	-.00022	.00099	.00008	-.00124	.00004

RUN NO. 31/ 0 RV/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
2.840	-4.376	3.02761	-20162	.14692	-.00100	.00057	-.03471	-.01667	-.15167	-.13524
2.840	-2.534	3.02112	-14422	.05638	-.00091	.00067	-.04891	-.01663	-.15166	-.13523
2.840	-1.551	3.01929	-11076	.14048	-.00082	.00060	-.04697	-.01651	-.15485	-.13520
2.840	-.511	3.02146	-.07989	.13884	-.00088	.00054	-.04877	-.01332	-.15485	-.13521
2.840	.532	3.01978	-.04891	.13685	-.00136	.00011	-.04594	.00265	-.15807	-.13521
2.840	1.555	3.01803	-.02010	.13500	-.00074	-.00030	-.04311	.00588	-.15806	-.13520
2.840	3.611	3.01906	.03704	.13188	-.00140	-.00060	-.04297	.01215	-.15807	-.13523
2.840	5.642	3.01780	.10401	.12769	-.00181	-.00113	-.04005	.00579	-.16128	-.13522
2.840	7.738	3.01531	.18276	.12419	-.00230	-.00092	-.03800	-.00377	-.16128	-.13522
2.840	11.871	3.01912	.29735	.11694	-.00379	-.00211	-.03687	-.00067	-.15808	-.12565
2.840	16.013	3.02384	.43777	.11027	-.00598	-.00511	-.03135	-.01338	-.16128	-.12243
2.840	20.147	3.02808	.58011	.10316	-.00609	-.00737	-.02569	.00260	-.16128	-.13202
2.840	24.301	3.02789	.74378	.09457	-.00680	-.00831	-.02349	-.03899	-.15487	-.13843
2.840	28.454	3.02719	.90289	.08670	-.00779	-.00789	-.02205	-.04860	-.14846	-.13524
GRADIENT	-.00093	.02891	-.00166	-.00131	-.00004	-.00017	.00140	.00429	-.00099	.00000

LA-6C, UPWT1040, ORBITER 0698 W/MOD. NOSE +OMS

REFERENCE DATA

SREF = 136.1808 50. IN. YMRP = 15.9636 INCHES
LREF = 8.9025 INCHES YMRP = .0000 INCHES
BREF = 17.9628 INCHES ZMRP = .0000 INCHES
SCALE = .0100 SCALE

PARAMETRIC DATA

BETA = .000 ELEVTR = -30.000
AILRON = .000 BDFLAP = -14.250
RUDDER = .000 RUDFLR = 40.000
K/L = 5.290 GT-LOC = 2.000

RUN NO. 34/ 0 RN/L = 1.49 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
1.900	-3.291	.00287	-2.7870	.20313	.09141	.00148	-.00391	.01006	-.11063	-.28265	-.23761
1.900	-1.440	.00356	-1.9320	.19978	.08102	.00153	-.00369	.00879	-.09661	-.28765	-.23983
1.900	-.411	.00156	-1.4829	.19838	.07372	.00158	-.00324	.00697	-.08337	-.29297	-.23983
1.900	.660	.00321	-.09913	.19582	.06801	.00125	-.00372	.00900	-.07594	-.29313	-.24007
1.900	1.668	.00292	-.05209	.19326	.06236	.00133	-.00345	.00838	-.07356	-.29322	-.23756
1.900	2.756	.00263	-.00515	.18999	.05615	.00162	-.00320	.00777	-.07072	-.29315	-.23746
1.900	4.855	.00287	.08491	.18325	.04467	.00170	-.00341	.00797	-.08648	-.29047	-.23741
1.900	6.955	.00251	.17487	.17562	.03454	.00133	-.00342	.00819	-.06272	-.27987	-.23478
1.900	9.036	.00327	.25930	.17045	.02820	.00102	-.00346	.00766	-.09904	-.27960	-.23706
1.900	13.221	.00557	.42813	.15862	.02196	.00087	-.00400	.00731	-.04341	-.27960	-.23441
1.900	17.439	.00627	.59973	.14909	.01877	.00047	-.00445	.00779	-.11758	-.28226	-.25296
1.900	20.532	.00790	.73013	.14257	.01850	.00007	-.00473	.00738	-.14405	-.29022	-.26355
GRADIENT	-.00001	.04472	-.00243	-.00576	.00001	.00006	-.00025	.00361	-.00096	.00019	.00019

RUN NO. 32/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
2.800	-4.393	.00101	-.22912	.15868	.07662	.00246	-.00462	.01362	-.02118	-.15323	-.11750
2.800	-2.575	.00092	-.18158	.15429	.07228	.00225	-.00502	.01483	-.01480	-.15324	-.11431
2.800	-1.561	.00235	-.14811	.15210	.07181	.00224	-.00460	.01220	-.00841	-.15324	-.11431
2.800	-.522	.00161	-.11472	.14946	.07045	.00185	-.00445	.01233	.00433	-.15645	-.11113
2.800	1.495	.00208	-.08604	.14724	.06886	.00208	-.00466	.01239	.01388	-.15645	-.10795
2.800	1.536	.00242	-.05525	.14440	.06577	.00202	-.00420	.01067	.02986	-.15645	-.10475
2.800	3.642	.01161	.01390	.13974	.06277	.00163	-.00439	.01180	.04585	-.16287	-.10155
2.800	5.669	.00174	.07826	.13512	.06020	.00135	-.00423	.01203	.06027	-.15966	-.10473
2.800	7.727	.00033	.14446	.13132	.05866	.00165	-.00397	.01136	.04268	-.16287	-.10793
2.800	11.963	.00036	.27668	.12348	.05803	.00080	-.00411	.01176	.01059	-.15967	-.11757
2.800	16.008	-.00164	.41891	.11700	.05821	.00081	-.00345	.01141	-.04379	-.16288	-.12718
2.800	20.144	-.00485	.56316	.10891	.05856	.00091	-.00242	.01113	-.05659	-.16289	-.13678
2.800	24.294	-.00770	.72176	.10034	.05933	.00115	-.00089	.00911	-.06622	-.15969	-.13679
2.800	28.468	-.00331	.86821	.09276	.06158	.00026	-.00086	.00589	-.07256	-.15008	-.13359
GRADIENT	.00012	.03035	-.00236	-.00168	-.00009	.00006	-.00038	.00899	-.00115	.00011	.00211

LA-8C, UPWT1040, ORBITER 0898 W/MOD. NOSE +ONS

(RP#208) (15 AUG 73)

REFERENCE DATA

REF = 136.1808 SQ.IN. XMRP = 13.9638 INCHES
 LREF = 9.9025 INCHES YMRP = .0000 INCHES
 BREF = 17.3628 INCHES ZMRP = .0000 INCHES
 SCALE = .0166 SCALE

PARAMETRIC DATA

BETA = 3.000 ELEVTR = -30.000
 AIRLON = .000 BDFLAP = -14.250
 RUDDER = .000 RUDFLR = 40.000
 K/L = 5.290 GT-LOC = 2.000

RUN NO. 35/ 0 RV/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
1.800	-3.296	3.03623	-2.7497	.20300	.09260	.00002	.00236	-.05867	-.04864	-.27979	-.23733
1.800	-1.435	3.03407	-1.9291	.19963	.08015	.00004	.00207	-.05466	-.02280	-.26775	-.23731
1.800	-.400	3.03221	-1.4197	.19799	.07408	-.00005	.00199	-.05300	-.01218	-.28773	-.23729
1.800	.658	3.03268	-.09517	.19532	.06693	-.00034	.00184	-.05293	-.00953	-.28308	-.23729
1.800	1.685	3.03156	-.05213	.19290	.06098	-.00053	.00133	-.05059	-.01483	-.28773	-.23729
1.800	2.764	3.03136	-.00502	.19037	.05610	-.00049	.00136	-.05046	-.02809	-.28773	-.23731
1.800	4.847	3.03250	.08693	.18313	.04332	-.00068	.00026	-.04810	-.17573	-.28245	-.23733
1.800	6.961	3.03502	.17905	.17642	.03405	-.00145	-.00163	-.04432	-.05692	-.27981	-.23999
1.800	9.048	3.03552	.26080	.16997	.02737	-.00180	-.00251	-.04189	-.08661	-.27195	-.23748
1.800	13.291	3.03716	.43119	.15868	.02309	-.00255	-.00311	-.04071	-.03655	-.26670	-.23754
1.800	17.448	3.04653	.60227	.14566	.01814	-.00387	-.00900	-.03033	-.13436	-.27204	-.24816
1.800	20.513	3.05371	.73282	.13907	.01797	-.00456	-.01298	-.02438	-.17920	-.27466	-.25606
GRADIENT		-.00066	.04443	-.00240	-.00599	-.00010	-.00024	.00123	-.00317	-.00022	-.00000

RUN NO. 33/ 0 RV/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
2.800	-4.396	3.02385	-.23312	.15888	.07511	-.00021	.00178	-.05443	-.05499	-.15329	-.11445
2.800	-2.362	3.01936	-.16412	.15336	.07365	.00000	.00179	-.05056	-.05541	-.15725	-.11435
2.800	-1.368	3.01665	-.13612	.15087	.07093	.00035	.00175	-.04775	-.07495	-.15322	-.11426
2.800	-.325	3.01770	-.11199	.14859	.06897	.00023	.00171	-.04863	.08457	-.15643	-.11425
2.800	.312	3.01596	-.07131	.14679	.06754	.00001	.00127	-.04575	.09410	-.15643	-.11107
2.800	1.550	3.01532	-.04501	.14384	.06660	.00003	.00113	-.04476	.11976	-.15322	-.10785
2.800	3.620	3.01642	.01954	.13938	.06319	.00031	.00008	-.04284	.12600	-.15644	-.10789
2.800	5.684	3.01628	.06621	.13518	.06077	-.00050	-.00018	-.03956	.13645	-.15645	-.12071
2.800	7.745	3.01551	.14985	.13184	.05979	-.00123	-.00020	-.04062	.14434	-.15966	-.12712
2.800	11.875	3.01695	.27762	.12383	.05878	-.00196	-.00119	-.03854	.150519	-.15965	-.12710
2.800	16.005	3.01868	.42107	.11565	.05731	-.00314	-.00442	-.03026	.16332	-.15963	-.13342
2.800	20.162	3.02282	.57346	.10789	.05633	-.00434	-.00699	-.02553	.17284	-.16284	-.13984
2.800	24.310	3.02527	.72535	.09950	.05534	-.00533	-.00812	-.02336	.18147	-.16643	-.13987
2.800	28.474	3.02822	.89170	.09220	.05474	-.00646	-.01061	-.02374	.18951	-.15322	-.13668
GRADIENT		-.00090	.03102	-.00240	-.00154	.00004	-.00021	.00141	-.01024	-.00037	-.00101

(RP6209) (15 AUG 75)

LA-8C, UPWT1040, ORBITER 089B W/MOD. NOSE +ONS

REFERENCE DATA

SREF = 136.1808 SQ.IN. XMRP = 15.9638 INCHES
 LREF = 8.9028 INCHES YMRP = .0000 INCHES
 BREF = 17.5628 INCHES ZMRP = .0000 INCHES
 SCALE = .0186 SCALE

PARAMETRIC DATA

BETA = .000 ELEVTR = -30.000
 AILRON = .000 BEFLAP = -14.250
 RUDDER = 10.000 RUDFLR = 40.000
 K/L = 5.290 GT-LOC = 2.000

RUN NO. 36/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	C/	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
1.900	-3.304	.01241	-.31012	.20690	.09927	.00583	-.01143	.02497	-.06366	-.20047	-.23281
1.900	-1.455	-.01198	-.22254	.20397	.08673	.00555	-.01132	.02446	-.06341	-.28304	-.23533
1.900	-.431	.01315	-.17359	.20205	.08120	.00542	-.01125	.02461	-.06077	-.28569	-.23798
1.900	.629	.01360	-.12684	.19983	.07477	.00490	-.01129	.02241	-.05813	-.28834	-.24062
1.900	1.669	.01186	-.07988	.19713	.06979	.00464	-.01117	.02333	-.05813	-.28834	-.24062
1.900	2.747	.01208	-.03102	.19454	.06240	.00486	-.01108	.02197	-.06341	-.28834	-.24062
1.900	4.620	.01110	.04080	.18935	.04967	.00462	-.01087	.02222	-.06451	-.28568	-.24060
1.900	6.914	.01235	.19055	.18150	.03882	.00384	-.01060	.02018	-.09243	-.28038	-.23796
1.900	9.023	.01320	.23813	.17530	.03060	.00315	-.01099	.02036	-.09240	-.28036	-.23794
1.900	13.219	.01410	.41029	.16431	.02231	.00227	-.01131	.02006	-.02897	-.27240	-.23793
1.900	17.411	.01550	-.57748	.15436	.01978	.00149	-.01176	.01976	-.08705	-.27504	-.24584
1.900	20.491	.01555	.70921	.14787	.01897	.00081	-.01163	.01943	-.12410	-.28301	-.25644
GRADIENT		-.00010	.04555	-.00228	-.00602	-.00016	.00008	-.00040	-.00191	-.00004	-.000102

RUN NO. 38/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
2.840	-4.372	.01057	-.22075	.16174	.08089	.00564	-.00984	.02052	.00077	-.16016	-.13399
2.840	-2.552	.01022	-.16613	.15680	.07817	.00573	-.00965	.01985	-.00529	-.15692	-.13391
2.840	-1.539	.01058	-.13746	.15438	.07634	.00515	-.00927	.01811	-.00228	-.16015	-.13395
2.840	-.517	.00930	-.10647	.15203	.07486	.00507	-.00931	.01917	-.00211	-.15692	-.13391
2.840	.532	.00916	-.07070	.14928	.07218	.00491	-.00940	.01927	.00111	-.16013	-.13391
2.840	1.573	.00904	-.03956	.14663	.07161	.00450	-.00947	.01936	.01065	-.16014	-.13393
2.840	3.607	.00935	.01765	.14185	.06752	.00418	-.00945	.01859	.02971	-.16336	-.13076
2.840	5.669	.00907	.08447	.13762	.06513	.00369	-.00956	.01879	.06799	-.16658	-.12758
2.840	7.734	.00815	.14567	.13386	.06227	.00304	-.00911	.01812	.06165	-.16337	-.12757
2.840	11.685	.00919	.28063	.12605	.06189	.00235	-.00913	.01670	.01997	-.16337	-.13399
2.840	16.008	.00606	.42087	.11888	.06014	.00149	-.00806	.01640	-.02472	-.16658	-.14037
2.840	20.159	.00568	.56780	.11012	.06066	.00069	-.00769	.01332	-.03754	-.16658	-.14678
2.840	24.306	.00347	.72719	.10138	.05974	.00046	-.00537	.01126	-.04392	-.16336	-.14357
2.840	28.467	.00358	.89124	.09379	.06262	-.00004	-.00417	.00823	-.07917	-.15373	-.14358
GRADIENT		-.00021	.03016	-.00248	-.00168	-.00020	.00004	-.00017	.00363	-.00047	.000031

LA-8C, LPWT1040, ORBITER 0898 W/MOD. NOSE +ONS

(RP6210) (15 AUG 73)

REFERENCE DATA

SREF = 136.1808 50.1N, XMRP = 15.9638 INCHES
 LREF = 9.9025 INCHES YMRP = .0000 INCHES
 BREF = 17.3626 INCHES ZMRP = .0000 INCHES
 SCALE = .0186 SCALE

PARAMETRIC DATA

BETA = 3.000 ELEVTR = -30.000
 ATLRON = .000 BDFLAP = -14.250
 RUGGER = 10.000 RUDFLR = 40.000
 K/L = 5.290 GT-LOC = 2.000

RUN NO. 37/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
1.900	-3.276	3.04695	-2.2631	.20416	.08643	.00397	-.00450	-.04513	-.03115	-.27224	-.23773
1.900	-1.442	3.04392	-2.0697	.20111	.08603	.00363	-.00495	-.04186	.01100	-.28025	-.23779
1.900	-.808	3.04152	-1.5769	.19535	.07974	.00361	-.00521	-.03943	.02400	-.28032	-.24052
1.900	-.559	3.04197	-1.1057	.19699	.07324	.00325	-.001942	-.03932	.02897	-.28305	-.24328
1.900	1.705	3.04016	-.06413	.19473	.06689	.00269	-.00555	-.03766	.01823	-.28310	-.24334
1.900	2.764	3.04235	-.01534	.19223	.06072	.00231	-.00592	-.03836	.01542	-.28314	-.24340
1.900	4.846	3.04160	.07237	.18611	.04830	.00174	-.00636	-.03666	.00206	-.28319	-.23819
1.900	6.937	3.04456	.16039	.18014	.04012	.00084	-.00794	-.03443	-.04804	-.28319	-.24083
1.900	9.036	3.04557	.24795	.17368	.03261	.00005	-.00912	-.03201	-.06647	-.28013	-.24081
1.900	13.226	3.04631	.41428	.16186	.02662	-.00131	-.00957	-.02938	-.00037	-.26459	-.23611
1.900	17.432	3.04533	.58546	.15056	.02308	-.00297	-.01482	-.02193	-.07685	-.26723	-.24073
1.900	20.496	3.04372	.71204	.14472	.02256	-.00417	-.01881	-.01679	-.15856	-.26982	-.24859
GRADIENT		-.00059	.04472	-.00220	-.00595	-.00029	-.00023	.00098	.00299	-.00095	-.00036

RUN NO. 39/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
2.860	-4.399	3.03079	-2.23039	.15930	.07657	.00324	-.00415	-.04337	.06487	-.16015	-.13397
2.860	-2.337	3.02806	-1.6580	.15428	.07283	.00333	-.00346	-.04302	.09507	-.16717	-.13400
2.860	-1.539	3.02512	-1.13220	.15191	.07146	.00310	-.00358	-.04007	.05801	-.16018	-.13405
2.860	-.497	3.02053	-.10127	.14948	.06910	.00295	-.00381	-.03534	.05494	-.16018	-.13403
2.860	.524	3.02443	-.07263	.14697	.06837	.00305	-.00416	-.03612	.07416	-.16338	-.13402
2.860	1.530	3.02100	-.04409	.14448	.06590	.00272	-.00442	-.03433	.09013	-.16338	-.13402
2.860	3.633	3.02092	.02282	.14005	.06440	.00204	-.00515	-.03237	.13184	-.16338	-.13400
2.860	5.664	3.02305	.07728	.13591	.06308	.00123	-.00523	-.03411	.10299	-.16659	-.13721
2.860	7.742	3.02169	.14574	.13249	.05953	.00061	-.00549	-.03203	.03244	-.16660	-.14044
2.860	11.866	3.02137	.27805	.12444	.05891	-.00077	-.00631	-.02895	-.01863	-.16660	-.14075
2.860	16.017	3.02432	.42075	.11663	.05822	-.00274	-.00884	-.02336	-.03434	-.16659	-.14682
2.860	20.152	3.02594	.56805	.10946	.05966	-.00407	-.01070	-.01853	-.02814	-.16659	-.14681
2.860	24.302	3.03002	.72502	.10155	.06027	-.00527	-.01212	-.01735	-.08888	-.16018	-.14682
2.860	26.470	3.03020	.89117	.09366	.06166	-.00641	-.01078	-.01946	-.05056	-.13657	-.14363
GRADIENT		-.00127	.03100	-.00239	-.00154	-.00014	-.00016	.00150	.00847	-.00053	-.00000

LA-8C, UPMT1040, ORBITER 0898 W/MOD. NOSE +OMS

(RP6211) (15 AUG 73)

REFERENCE DATA

SREF = 136.1808 SQ. IN. XMRP = 15.9638 INCHES
 LREF = 9.925 INCHES YMRP = .0000 INCHES
 BREF = 17.3628 INCHES ZMRP = .0000 INCHES
 SCALE = .0168 SCALE

PARAMETRIC DATA

BETA = .000 ELEVTR = -10.000
 AIRLON = 10.000 BDFLAP = -14.250
 RUDDER = .000 RUDDFLR = 40.000
 K/L = 5.290 GT-LOC = 2.000

RUN NO. 24/ D RN/L = 1.49 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
1.900	-3.249	-0.00477	-1.9348	.17931	.04465	.01186	.00065	.00461	-.14785	-.27832	-.25173
1.900	-1.405	-0.00326	-1.0920	.17801	.03456	.01166	-.00019	.00476	-.12854	-.28068	-.25134
1.900	-3.36	-0.00232	-0.6383	.17696	.03015	.01174	-.00043	.00409	-.12542	-.28316	-.24843
1.900	.699	-0.00765	-0.1650	.17603	.02367	.01153	-.00110	.00415	-.12272	-.28851	-.24574
1.900	1.746	-0.00302	.02906	.17456	.01923	.01129	-.00151	.00655	-.12032	-.29396	-.24589
1.900	2.774	-0.00146	.07189	.17234	.01429	.01128	-.00188	.00579	-.12300	-.29496	-.24471
1.900	4.674	-0.00112	.15872	.16894	.00311	.01116	-.00269	.00666	-.14075	-.28255	-.23796
1.900	6.979	.00037	.24455	.16334	-.00445	.01069	-.00350	.00682	-.13780	-.27977	-.24039
1.900	9.052	.00186	.32934	.16013	-.01151	.01016	-.00429	.00699	-.12332	-.27919	-.24224
1.900	13.239	.00472	.49552	.15317	-.01904	.00986	-.00595	.00734	-.08245	-.28136	-.24690
1.900	17.442	.00514	.66854	.14487	-.02263	.00946	-.00724	.00856	-.11623	-.28909	-.25983
1.900	20.520	.00734	.80385	.13643	-.02766	.00910	-.00809	.00735	-.11620	-.27842	-.25185
GRADIENT		.00040	.04340	-.00148	-.00507	-.00041	-.00030	.00030	.00097	-.00130	.00166

RUN NO. 19/ D RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
2.840	-4.391	.00013	-1.7329	.14318	.04957	.00782	-.00242	.00953	.00865	-.15210	-.11649
2.840	-2.562	.00044	-1.1372	.14009	.04799	.00735	-.00253	.00880	-.01357	-.15207	-.11643
2.840	-1.532	-.00077	-.09460	.13864	.04710	.00715	.00262	.00980	-.02000	-.15529	-.11964
2.840	-.514	.00034	-.06597	.13695	.04536	.00710	-.00311	.00982	-.02318	-.15529	-.11963
2.840	.526	-.00092	-.03007	.13558	.04521	.00649	-.00320	.01087	-.02640	-.15529	-.12284
2.840	1.564	-.00113	-.00147	.13393	.04276	.00669	-.00297	.01007	-.02645	-.15529	-.12286
2.840	3.611	.00043	.05337	.13072	.04101	.00609	-.00339	.00925	-.02342	-.15853	-.11932
2.840	5.668	-.00099	.11785	.12721	.03846	.00605	-.00356	.01039	-.00100	-.15531	-.11011
2.840	7.733	-.00013	.18164	.12464	.03669	.00593	-.00380	.00965	-.01055	-.15531	-.11329
2.840	11.875	-.00005	.31661	.11825	.03379	.00586	-.00475	.01093	-.02653	-.15530	-.11969
2.840	16.006	-.00079	.45681	.11373	.03207	.00594	-.00487	.01050	-.02973	-.15852	-.13249
2.840	20.145	.00017	.60825	.10563	.02798	.00625	-.00536	.00911	-.02977	-.15852	-.13251
2.840	24.303	.00045	.77702	.09809	.02367	.00645	-.00567	.00781	-.05854	-.15852	-.13570
2.840	28.474	.00067	.94801	.09168	.02218	.00691	-.00619	.00653	-.06179	-.14690	-.13252
GRADIENT		-.00005	.02826	-.00154	-.00310	-.00021	-.00013	.00006	-.00372	-.00076	-.00007

LA-6C, UPWT1040, ORBITER 0898 W/MOD, NOSE +OMS

(RP6212) (15 AUG 73)

REFERENCE DATA

SREF = 136.1608 SQ.IN. XMRP = 15.9638 INCHES
 LREF = 8.9025 INCHES YMRP = .0000 INCHES
 BREF = 17.3628 INCHES ZMRP = .0000 INCHES
 SCALE = .0188 SCALE

PARAMETRIC DATA

BETA = 3.000 ELEVTR = -10.000
 AILRON = 10.000 BDFLAP = -14.250
 RUDDER = .000 RUDFLR = 40.000
 K/L = 5.290 GT-LOC = 2.000

RUN NO. 25/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CFB1	CFB2	CPC
1.900	-3.250	3.02614	-1.1276	.17797	.04288	.00975	.00521	-.05600	-.15668	-.28139	-.24429
1.900	-1.417	3.02905	-1.1223	.17607	.03442	.00940	.00381	-.05209	-.14614	-.28408	-.24168
1.900	-.352	3.02497	-.06325	.17529	.02885	.00908	.00336	-.05124	-.13290	-.28142	-.24168
1.900	-.666	3.02498	-.01826	.17409	.02226	.00899	.00265	-.04967	-.12496	-.28408	-.24168
1.900	1.733	3.02560	.02493	.17276	.01771	.00881	.00179	-.04412	-.11443	-.28144	-.24171
1.900	2.777	3.02550	.06820	.17118	.01388	.00877	.00136	-.04727	-.10368	-.27890	-.24173
1.900	4.859	3.02719	.15433	.16796	.00269	.00845	.00006	-.04567	-.10923	-.27887	-.24441
1.900	6.978	3.02973	.24454	.16489	-.00604	.00800	-.00165	-.04335	-.11197	-.28416	-.24973
1.900	9.039	3.03066	.32843	.16295	-.01112	.00716	-.00296	-.04096	-.11470	-.28420	-.25508
1.900	13.253	3.03341	.49494	.15226	-.01789	.00540	-.00551	-.03697	-.09004	-.28153	-.25505
1.900	17.436	3.04557	.67040	.14147	-.02455	.00486	-.01202	-.02892	-.11993	-.27887	-.25505
1.900	20.527	3.05163	.80105	.13276	-.02614	.00400	-.01644	-.02219	-.15690	-.26557	-.24973
GRADIENT		.00014	.04281	-.00122	-.00496	-.00016	-.00063	.00125	.017681	.00050	-.00002

RUN NO. 20/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CFB1	CFB2	CPC
2.860	-4.385	3.01745	-1.16623	.14264	.04901	.00461	.00347	-.05239	.00548	-.15209	-.12288
2.860	-2.560	3.01798	-.12049	.13960	.04721	.00434	.00278	-.05140	-.02333	-.13209	-.12289
2.860	-1.537	3.01733	-.06942	.13788	.04566	.00407	.00256	-.05037	-.02981	-.15211	-.12612
2.860	-.504	3.01677	-.06063	.13623	.04363	.00399	.00200	-.04851	-.03612	-.15209	-.12609
2.860	.525	3.03397	-.02727	.13494	.04359	.00399	.00149	-.04566	-.03613	-.15531	-.12609
2.860	1.539	3.03333	.00142	.13331	.04284	.00385	.00128	-.04467	-.03612	-.15530	-.12609
2.860	3.622	3.03326	.06362	.13079	.04175	.00357	.00051	-.04270	-.03614	-.15531	-.12930
2.860	5.864	3.03267	.12565	.12746	.03979	.00282	-.00037	-.03982	-.02659	-.15832	-.13251
2.860	7.739	3.03193	.17982	.12449	.03801	.00248	-.00056	-.03876	-.02654	-.15531	-.13250
2.860	11.862	3.03358	.31963	.11812	.03550	.00165	-.00283	-.03399	-.02971	-.15851	-.13249
2.860	16.014	3.03634	.45736	.11228	.03440	.00145	-.00427	-.02884	-.02659	-.15852	-.13251
2.860	20.161	3.04231	.61885	.10556	.03035	.00103	-.00976	-.02286	-.02339	-.16173	-.13251
2.860	24.302	3.04590	.77507	.09736	.02504	-.00017	-.01169	-.02168	-.04907	-.15833	-.13254
2.860	28.470	3.04748	.94915	.09051	.02291	-.00057	-.01266	-.02125	-.05541	-.14890	-.12932
GRADIENT		.00263	.02918	-.00149	-.00094	-.00015	-.00038	.00135	-.00461	-.00054	-.00076

LA-8C, JPWT1040, GRB.TER 0898 W/MOD. NOSE +OMS

(RP6213) (15 AUG 73)

REFERENCE DATA

SREF = 136.1608 SQ.IN. YMRP = 15.9638 INCHES
 LREF = 8.9025 INCHES YMRP = .0000 INCHES
 BREF = 17.5628 INCHES ZMRP = .0000 INCHES
 SCALE = .0186 SCALE

PARAMETRIC DATA

ALPHA = 15.000 ELEVTR = -10.000
 AILRON = 10.000 BDFLAP = -14.250
 RUDDER = .000 RUDFLR = 40.000
 K/L = 5.290 GT-LOC = 2.000

RUN NO. 26/ 0 RM/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ALPHA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
1.900	-6.174	16.40833	.62574	.14201	-.02642	.01957	-.03277	.09640	-.08304	-.29749	-.25247
1.900	-4.066	16.41145	.62779	.14303	-.02430	.01576	-.00258	.06209	-.06217	-.27902	-.24730
1.900	-2.043	16.41821	.62486	.14584	-.02189	.01267	-.00518	.03440	-.08601	-.28169	-.24398
1.900	-1.018	16.41993	.62412	.14678	-.02272	.01107	-.00641	.02173	-.10455	-.28437	-.25529
1.900	.006	16.42245	.62555	.14684	-.02127	.00932	-.00702	.00761	-.10722	-.28703	-.25530
1.900	1.033	16.42740	.62513	.14677	-.02141	.00764	-.00766	-.00651	-.10977	-.29700	-.25790
1.900	2.017	16.41730	.62474	.14578	-.02222	.00640	-.00863	-.01835	-.09915	-.28432	-.25787
1.900	4.110	16.41084	.62369	.14224	-.02387	.00319	-.01164	-.04535	-.11758	-.26839	-.25520
1.900	6.141	16.41114	.62255	.14158	-.02467	-.00016	-.01200	-.07818	-.12800	-.26832	-.26306
1.900	8.210	16.40167	.61992	.13996	-.02744	-.00418	-.00637	-.12274	-.13343	-.27369	-.26577
GRADIENT	-.007402	-.00036	-.00007	-.00005	-.00005	-.00154	-.00104	-.01314	-.00591	.00082	-.00116

MACH	BETA	ALPHA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
2.860	-6.135	14.42818	.41117	.11370	.02932	.01617	-.00729	.09967	.00879	-.16493	-.12606
2.860	-4.065	14.44901	.41208	.11470	.03177	.01293	-.00544	.06691	-.02010	-.16494	-.12928
2.860	-2.017	14.45125	.40583	.11542	.03357	.00941	-.00407	.03502	-.03932	-.16494	-.13369
2.860	-1.039	14.46372	.41259	.11703	.03578	.00783	-.00418	.02368	-.03295	-.16494	-.13570
2.860	-.990	14.46958	.41424	.11707	.03582	.00592	-.00434	.06349	-.03229	-.16494	-.13571
2.860	.990	14.47136	.41349	.11497	.03482	.00416	-.00487	-.00288	-.03619	-.16494	-.13571
2.860	2.018	14.47568	.41052	.11451	.03530	.00234	-.00464	-.01702	-.03296	-.16494	-.13570
2.860	4.077	14.46971	.41139	.11332	.03519	-.00102	-.00389	-.04897	-.02339	-.15852	-.13251
2.860	6.092	14.46542	.40960	.11225	.03228	-.00434	-.00149	-.08258	-.01377	-.15531	-.13250
2.860	8.166	14.46418	.41252	.11051	.03064	-.00770	.00045	-.11807	.01816	-.15853	-.13572
GRADIENT	.00284	.00017	-.00019	-.00038	-.00005	-.00173	.00010	-.01393	-.00008	.00061	-.00030

RUN NO. 21/ 0 RM/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

LA-8C, UPWT1040, ORBITER 0898 //MOD, NOSE +OMS

(RP6214) (15 AUG 73)

REFERENCE DATA

SREF = 136.1808 SQ. IN. XMRP = 15.5638 INCHES
 I.REF = 8.9025 INCHES YMRP = .0000 INCHES
 BREF = 17.5628 INCHES ZMRP = .0000 INCHES
 SCALE = .0168 SCALE

PARAMETRIC DATA

ALPHA = 20.000 ELEVTR = -10.000
 AIRLON = 10.000 BOFLAP = -14.250
 RUDDER = .000 RUDFLR = 40.000
 K/L = 5.290 GT-LOC = 2.000

RUN NO. 27/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	Y' / MA	ON	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
1.900	-6.169	20.51100	.83	.12785	-.05626	.02098	.00302	.08705	-.11338	-.28712	-.25278
1.900	-4.097	20.51713	.87094	.13135	-.03183	.01696	.00209	.05265	-.09949	-.28180	-.24747
1.900	-2.049	20.52234	.80225	.13386	-.02976	.01305	-.00263	.02926	-.12851	-.27121	-.25011
1.900	-1.002	20.52736	.60243	.13570	-.02703	.01123	-.00539	.01864	-.12051	-.27382	-.25007
1.900	-.011	20.52920	.80019	.13635	-.02660	.00916	-.00823	.00006	-.11256	-.27381	-.25005
1.900	1.116	20.52104	.79401	.13593	-.02491	.00745	-.01101	-.00177	-.12579	-.27117	-.25007
1.900	2.066	20.51974	.79345	.13445	-.02641	.00563	-.01413	-.01220	-.15220	-.26854	-.25008
1.900	4.123	20.52196	.79832	.13145	-.02837	.00213	-.01670	-.03752	-.15223	-.26855	-.24748
1.900	6.136	20.51419	.79450	.12872	-.03104	-.00192	-.01773	-.07304	-.14956	-.27383	-.25008
1.900	8.227	20.51270	.78644	.12800	-.02854	-.00702	-.01459	-.11620	-.14434	-.27651	-.25539
GRADIENT	.00017		-.00071	.00004	.00052	-.00181	-.00259	-.01079	-.00162	.00141	.00000

RUN NO. 22/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ALPHA	ON	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
2.860	-6.149	20.15609	.62109	.10243	.02378	.01860	-.00132	.08971	-.02983	-.16174	-.13892
2.860	-4.071	20.15234	.61511	.10385	.02558	.01471	-.00145	.05622	-.01380	-.16494	-.13891
2.860	-2.041	20.14037	.61620	.10518	.02893	.01006	-.00106	.02860	-.02680	-.16173	-.13571
2.860	-1.039	20.17209	.61827	.10536	.02900	.00843	-.00315	.01807	-.02661	-.16174	-.13572
2.860	.001	20.17287	.61762	.10548	.03051	.00640	-.00311	.00824	-.02986	-.16174	-.13573
2.860	1.003	20.16906	.61258	.10621	.03168	.00407	-.00685	-.00154	-.02984	-.16174	-.13572
2.860	2.004	20.17482	.61680	.10616	.03114	.00223	-.00882	-.01132	-.01408	-.15853	-.13573
2.860	4.082	20.17075	.61542	.10438	.03080	-.00202	-.00887	-.03681	-.02026	-.15853	-.13572
2.860	6.121	20.17265	.61151	.10254	.03020	-.00432	-.00813	-.07088	.00856	-.15532	-.13572
2.860	8.136	20.17128	.61010	.10078	.02901	-.00358	-.00799	-.09929	.01820	-.15531	-.13251
GRADIENT	.00232		-.00006	.00011	.00064	-.00203	-.00115	-.01123	-.00023	.00075	.00000

DATE 29 JAN 74 TABULATED SOURCE DATA FOR LA-6C (LARC UPWT 1040)

(RF0215) (15 AUG 73)

LA-6C, UPWT1040, ORBITER 0698 W/MOD. NOSE +ONS

REFERENCE DATA
 SREF = 136.1808 SQ. IN. XMRP = 15.9636 INCHES ALPHA = 25.000 ELEVTR = -10.000
 LREF = 6.9025 INCHES YMRP = .0000 INCHES AILRON = 10.000 BDFLAP = -14.250
 BREF = 17.5628 INCHES ZMRP = .0000 INCHES RUDDER = .000 RUDFLR = 40.000
 SCALE = .0188 SCALE K/L = 5.290 GT-LOC = 2.000

PARAMETRIC DATA

RUN NO. 23/ 0 RN/L = 1.50 GRADIENT INTERVAL = -.500/ 5.00

MACH	BETA	ALPHA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
2.000	-6.132	25.37732	.81968	.09396	.02222	.02090	.00199	.07486	-.03622	-.15532	-.13892
2.000	-4.075	25.37603	.81829	.09363	.02354	.01973	.00123	.04612	-.05541	-.15532	-.13892
2.000	-2.024	25.38730	.82173	.09456	.02451	.01108	-.00038	.02504	-.07462	-.15532	-.13892
2.000	-1.042	25.38544	.81900	.09533	.02505	.00900	-.00230	.01616	-.07462	-.15532	-.13892
2.000	.000	25.39167	.82362	.09561	.02536	.00657	-.00351	.00709	-.07141	-.15532	-.13892
2.000	1.003	25.38739	.82091	.09565	.02508	.01446	-.00820	-.00098	-.06182	-.15211	-.13892
2.000	2.005	25.38625	.82072	.09577	.02586	.00252	-.01076	-.00899	-.06182	-.15211	-.13573
2.000	4.068	25.38565	.81911	.09420	.02636	-.00298	-.01287	-.03314	-.04908	-.15212	-.13254
2.000	6.123	25.38712	.81906	.09477	.02534	-.00830	-.01337	-.05610	-.03951	-.14892	-.13255
2.000	8.181	25.39083	.81652	.09524	.02645	-.01298	-.01291	-.08737	-.04582	-.15211	-.13252
	GRADIENT	.00070	.00008	.00012	.00033	-.00226	-.00195	-.00962	.00149	.00052	.00075

LA-8C, UPWT1040, ORBITER 0898 W/MOD. NOSE +OMS (RP0216) (15 AUG 73)

REFERENCE DATA

SRZF = 136.1808 SQ. IN. ZMRP = 15.9638 INCHES
 LRZF = 8.9025 INCHES YMRP = .0000 INCHES
 BRZF = 17.5428 INCHES ZMRP = .0000 INCHES
 SCALE = .0188 SCALE

PARAMETRIC DATA

ALPHA = 15.000 ELEVTR = -10.000
 AIRLON = .000 BDFLAP = -14.250
 RUDDER = .000 RUOFLR = 40.000
 K/L = 5.280 GT-LOC = 2.000

RUN NO. 12/ D RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ALPHA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
1.900	-6.172	16.41833	.67927	.13779	-.02585	.01047	-.00074	.09702	-.06359	-.26245	-.24018
1.900	-4.104	16.42130	.62544	.13957	-.02460	.00674	-.00073	.06270	-.07372	-.27193	-.24028
1.900	-2.042	16.42373	.63731	.14104	-.02395	.00365	-.00312	.03584	-.07379	-.26665	-.24295
1.900	-1.034	16.42417	.63389	.14185	-.02337	.00203	-.00422	.02243	-.07903	-.26664	-.24030
1.900	1.032	16.42572	.62932	.14183	-.02422	.00028	-.00484	.00830	-.06576	-.26926	-.23762
1.900	2.019	16.42684	.63098	.14148	-.02347	-.00134	-.00529	-.00501	-.07094	-.26922	-.23757
1.900	4.090	16.42464	.62870	.14063	-.02305	-.00271	-.00646	-.01845	-.07880	-.26655	-.23754
1.900	6.143	16.41818	.63155	.13878	-.02433	-.00572	-.00946	-.04386	-.11581	-.25595	-.24284
1.900	8.152	16.42105	.63199	.13775	-.02519	-.00966	-.01017	-.07750	-.12902	-.26656	-.25341
1.900	GRADIENT	16.41589	.63330	.13532	-.03022	-.01352	-.01668	-.12054	-.12910	-.26130	-.25546
		-.00019	.00015	-.00010	.00006	-.00153	-.00099	-.01309	-.00397	.00143	-.00008

RUN NO. 16/ D RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ALPHA	CN	CA	CLM	CBL	CYN	CY	CPB1	CPB2	CPC
2.860	-6.160	14.97584	.42353	.10968	.02601	.01126	-.00385	.10428	.03632	-.15930	-.12394
2.860	-3.990	14.98503	.42433	.11064	.02757	.00787	-.00375	.06691	.00752	-.15930	-.12394
2.860	-2.001	14.98561	.41369	.11185	.03150	.00389	-.00284	.03870	.01075	-.15930	-.12713
2.860	-1.036	15.00165	.42509	.11240	.03157	.00244	-.00309	.02455	.02034	-.15930	-.13033
2.860	1.021	14.99226	.42660	.11258	.03077	.00057	-.00355	.01222	.03306	-.15931	-.13035
2.860	2.016	14.99375	.42366	.11194	.03124	-.00154	-.00360	-.00107	.03950	-.15930	-.12714
2.860	4.096	14.99606	.42767	.11194	.03153	-.00304	-.00376	-.01334	.02020	-.15931	-.12397
2.860	6.091	14.99340	.42638	.11027	.02885	-.00673	-.00358	-.04352	-.02773	-.15931	-.12717
2.860	8.149	14.98461	.42172	.10380	.02634	-.01043	-.00111	-.07807	-.01813	-.15630	-.13356
2.860	GRADIENT	14.98964	.43178	.10723	.02277	-.01380	-.00045	-.11457	-.00214	-.15931	-.13675
		.00111	.00101	-.00004	.00011	-.00180	-.00004	-.01351	-.00248	-.00000	-.00007

LA-8C, UPVT1040, ORBITER 0898 W/MOD. NOSE *OMS

(RP0217) (15 AUG 73)

REFERENCE DATA

SREF = 136.1808 SQ.IN. XMRP = 15.9638 INCHES
 LREF = 8.9025 INCHES YMRP = .0000 INCHES
 BREF = 17.5628 INCHES ZMRP = .0000 INCHES
 SCALE = .0188 SCALE

PARAMETRIC DATA

ALPHA = 20.000 ELEVTR = -10.000
 AIRLON = .000 BDFLAP = -14.250
 RUDDER = .000 RUDFLR = 40.000
 K/L = 3.290 GT-LOC = 2.000

RUN NO. 13/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ALPHA	CN	CA	CLM	CBL	CYN	CY	CFB1	CFB2	CPC
1.900	-6.191	20.52263	.60448	.12329	-.03532	.01245	.00610	.08666	-.12130	-.27195	-.24295
1.900	-4.079	20.52690	.60434	.12653	-.03340	.00805	.00585	.05195	-.11353	-.27467	-.24040
1.900	-2.049	20.53585	.60618	.12987	-.02849	.00421	.00095	.02856	-.11616	-.26672	-.24040
1.900	-1.059	20.55569	.60412	.13200	-.02807	.00209	-.00184	.01872	-.10816	-.26668	-.24299
1.900	-.013	20.5734	.60765	.13297	-.02852	.00008	-.00354	.00813	-.12135	-.26933	-.24563
1.900	.695	20.53742	.60555	.13238	-.02806	-.00137	-.00684	-.00006	-.11341	-.26667	-.24297
1.900	2.024	20.53330	.60480	.13055	-.02888	-.00369	-.00996	-.01151	-.10822	-.26141	-.23774
1.900	4.160	20.52802	.60576	.12674	-.03031	-.00736	-.01499	-.03649	-.15046	-.25348	-.23512
1.900	6.157	20.52246	.60360	.12438	-.03210	-.01175	-.01471	-.07230	-.16638	-.26413	-.24310
1.900	8.247	20.52979	.79894	.12346	-.03207	-.01695	-.01166	-.11463	-.15335	-.26686	-.24847
GRADIENT	.00000	.00004	.00008	.00014	.00026	-.00168	-.00256	-.01052	-.00325	.00223	.00063

RUN NO. 17/ 0 RN/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

MACH	BETA	ALPHA	CN	CA	CLM	CBL	CYN	CY	CFB1	CFB2	CPC
2.600	-6.123	20.13296	.60728	.09992	.02314	.01259	.00091	.08819	-.01496	-.15631	-.13996
2.600	-4.092	20.14266	.61079	.10117	.02323	.00895	.00082	.05803	-.01811	-.15630	-.13995
2.600	-2.042	20.14623	.60460	.10244	.02585	.00444	.00091	.02962	-.02134	-.15631	-.13996
2.600	-1.039	20.15765	.61602	.10325	.02676	.00200	-.00089	.01892	-.02136	-.15631	-.13996
2.600	-.002	20.15530	.61085	.10344	.02421	-.00006	-.00255	.01009	-.02772	-.15951	-.13995
2.600	.900	20.15781	.61519	.10362	.02740	-.00168	-.00439	.00035	-.02458	-.15952	-.13677
2.600	2.003	20.14936	.60993	.10379	.02684	-.00410	-.00656	-.00952	-.01817	-.15631	-.13676
2.600	4.099	20.14741	.61083	.10178	.02499	-.00843	-.00668	-.03702	-.02133	-.15630	-.13676
2.600	6.083	20.13935	.60435	.09988	.02417	-.01296	-.00601	-.07005	-.00854	-.15309	-.13675
2.600	8.158	20.14767	.61202	.09825	.02039	-.01705	-.00574	-.10027	-.00217	-.15310	-.13356
GRADIENT	.00059	.00013	.00023	.00022	.00021	-.00113	-.00113	-.01113	-.00023	-.00007	.00052

LA-8C, UPWT1040, ORBITER 0898 W/MOD. NOSE +ONS

(RP6218) (15 AUG 75)

REFERENCE DATA

SREF = 136.1808 50.1IN, XMRP = 15.9634 INCHES
 LREF = 8.9025 INCHES YMRP = .0000 INCHES
 BREF = 17.9628 INCHES ZMRP = .0000 INCHES
 SCALE = .0188 SCALE

ALPHA = 25.000 ELEVTR = -10.000
 AILRON = .000 BDFLAP = -14.250
 RUDDER = .000 RUOFLR = 40.000
 K/L = 5.290 GT-LOC = 2.000

RUN NO. 18/ 0 RM/L = 1.50 GRADIENT INTERVAL = -5.00/ 5.00

PARAMETRIC DATA

MACH	BETA	ALPHA	CN	CA	CLM	CBL	CYN	CY	CFB1	CFB2	CPC
2.800	-6.151	25.36008	.81252	.09129	.01870	.01426	.00555	.07591	-.03091	-.15309	-.13995
2.800	-4.078	25.36733	.81617	.09099	.02048	.00964	.00485	.04912	-.01811	-.15309	-.13675
2.800	-2.046	25.36667	.81235	.09234	.02248	.00433	.00332	.02599	-.05971	-.14989	-.13676
2.800	-1.060	25.37363	.81614	.09309	.02277	.00170	.00114	.01517	-.06617	-.15311	-.13679
2.800	.018	25.37117	.81642	.09340	.02270	.00034	-.00778	.00710	-.07249	-.15310	-.13676
2.800	1.002	25.37075	.81608	.09393	.02262	-.03244	-.00448	-.00095	-.06610	-.15310	-.13996
2.800	2.002	25.36674	.81336	.09350	.02318	-.01443	-.00677	-.00800	-.05653	-.14990	-.13997
2.800	4.065	25.36848	.81477	.09208	.02221	-.00964	-.00901	-.03215	-.03411	-.14989	-.13675
2.800	6.106	25.36557	.81317	.09205	.02009	-.01547	-.00983	-.05805	-.02449	-.14988	-.13355
2.800	8.163	25.36356	.81639	.09172	.01929	-.02012	-.00971	-.08745	-.03088	-.14988	-.13355
	GRADIENT	-.00005	-.00008	.00017	.00019	-.00231	-.00190	-.00956	-.00133	.00030	-.00022

NASA-MSFC-MAF