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# A MULTILEVEL CONTROL SYSTEM FOR THE LARGE SPACE TELESCOPE

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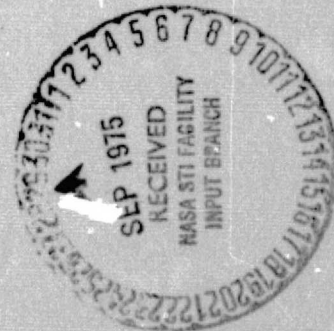
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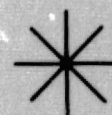
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## 1. INTRODUCTION

The principal objective of this report is to outline a multilevel control scheme for the Large Space Telescope (LST). The concept and methodology of the scheme is based upon the decomposition-aggregation stability analysis of large-scale systems [1-3], which was used to study structural properties of the control system for a spinning flexible spacecraft [4, 5].

The two-level analysis of the decomposition-aggregation method is ideally suitable for designing a multilevel feedback control [6-10] for dynamic systems composed of interconnected subsystems. Local controllers on the subsystem level are used to stabilize (or optimize) the decoupled subsystems. On the second hierarchical level the global controllers are used to minimize the interactions among the subsystems, and make the control system meet the required performance characteristics for the overall system. This multilevel strategy can solve complex control problems "piece-by-piece" and make the computer use attractive in cases when the direct approach is either not feasible (excessive computer storage), or it is uneconomical (excessive computer time).

The detailed plan of the report is as follows:

In Section 2, we will develop a nonlinear model for the LST which is based upon the linear model described in [11]. The nonlinear representation will serve as a realistic model for evaluating the potentials of the multilevel schemes for control of the LST.

In Section 3, we will outline the general multilevel stabilization algorithm [6-8]. Both local and global controllers are involved. The local controllers are used to stabilize each decoupled subsystem by any of the classical techniques such as pole-shifting, root-locus, parameter plane, etc. The role of the global controllers is to minimize the effect of interactions among the subsystems. Finally, the aggregate system is constructed on the higher hierarchical level to conclude stability of the overall composite system. We

will consider a class of dynamic systems [12] which can always be stabilized by the proposed scheme using local controllers only. Since the LST model developed in Section 2 is in that class, we will be able to effectively design the feedback control which stabilizes the LST.

In Section 4, we will present a multilevel optimization scheme for control of large-scale systems [9, 10]. The local controllers are used to optimize the decoupled subsystems with respect to quadratic cost. The global controllers are applied to reduce the subsystem interactions, or entirely decouple the subsystems as is the case of the LST. While this control scheme results in a suboptimal performance when the effective interactions are present, it produces an optimal control when the total decoupling takes place. Thus, the design procedure can effectively be used for constructing an optimal control system for the LST.

Both the stabilization and the optimization multilevel schemes are entirely computerized. The description of the programs is provided in the Appendix.

This report is written under the supervision and with the participation of the Principal Investigator, D. D. Šiljak. Investigator S. K. Sundareshan developed the model of LST in Section 2, and Sections 4 and A.2 on multilevel optimization. Investigator M. B. Vukčević developed the multilevel stabilization scheme presented in Sections 3 and A.1.

## 2. DEVELOPMENT OF A MODEL FOR THE LST

The Large Space Telescope (LST) is modeled as a rigid body with three orthogonally mounted reaction wheel actuators and is considered to be subject to gravitational and magnetic disturbance torques. Unlike in the earlier analyses [11], nonlinear coupling phenomena are not ignored and a complete three-axes model for the spacecraft is obtained as a nonlinear interconnected system. The interconnections represent the coupling between the motions along the individual axes. Hence this model will be a more accurate description of the LST, which however, is necessary due to the precision pointing requirements demanded of the control system.

The spacecraft's equation of motion can be written down from the Euler equations [11], as

$$I \cdot \dot{\omega} + \omega \times I \cdot \omega + \sum_{i=1}^3 \{ \omega \times \omega_i \operatorname{tr} I_i + 2\omega_i \times I_i \cdot \omega + I_i \cdot \dot{\omega}_i + \omega_i \times I_i \cdot \omega_i \} = M \quad (2.1)$$

and

$$I_i \cdot \dot{\omega}_i + \omega \times I_i \cdot \omega + \omega \times \omega_i \operatorname{tr} I_i + 2\omega_i \times I_i \cdot \omega + I_i \cdot \dot{\omega}_i + \omega_i \times I_i \cdot \omega_i = M_i, \quad i = 1, 2, 3 \quad (2.2)$$

where

$I$  is the inertia tensor of the LST given by

$$I = \begin{bmatrix} I_x & 0 & 0 \\ 0 & I_y & 0 \\ 0 & 0 & I_z \end{bmatrix},$$

$I_x, I_y, I_z$  denoting the components along the three axes constituting an inertial reference frame  $I_{rf}$ ;

$I_i, i = 1, 2, 3$ , are the inertia tensors of the three reaction wheels that are mounted orthogonally and parallel to the

axes constituting the standard body-fixed reference frame  $B_{rf}$  and hence can be expressed as

$$I_1 = \begin{bmatrix} I_{1x} & 0 & 0 \\ 0 & I_{1y} & 0 \\ 0 & 0 & I_{1y} \end{bmatrix}; \quad I_2 = \begin{bmatrix} I_{2z} & 0 & 0 \\ 0 & I_{2y} & 0 \\ 0 & 0 & I_{2z} \end{bmatrix};$$

$$I_3 = \begin{bmatrix} I_{3x} & 0 & 0 \\ 0 & I_{3x} & 0 \\ 0 & 0 & I_{3z} \end{bmatrix}$$

$\omega$  is the angular velocity vector of the LST relative to the frame  $I_{rf}$ ;

$\omega_i$ ,  $i = 1, 2, 3$ , are the angular velocity vectors of the reaction wheels relative to the frame  $I_{rf}$ ;

$M$  is the total external torque acting on the LST; and

$M_i$ ,  $i = 1, 2, 3$ , are the internal torques on the reaction wheels.

The angular velocity  $\omega$  can be expressed in terms of the rates of angular deviations along the three axes of  $I_{rf}$  as

$$\omega = \begin{bmatrix} \dot{\phi} \\ \dot{\theta} \\ \dot{\psi} \end{bmatrix} \quad (2.3)$$

where  $\phi$  is the roll angle,  $\theta$  is the pitch angle and  $\psi$  is the yaw angle. Similarly the angular velocities  $\omega_i$  of the reaction wheels can be expressed

in terms of the components as

$$\omega_1 = \begin{bmatrix} v_1 \\ 0 \\ 0 \end{bmatrix}; \quad \omega_2 = \begin{bmatrix} 0 \\ v_2 \\ 0 \end{bmatrix}; \quad \omega_3 = \begin{bmatrix} 0 \\ 0 \\ v_3 \end{bmatrix}. \quad (2.4)$$

Equations (2.1) and (2.2) can now be simplified into the following four sets of scalar equations:

(i) Equations governing the motion of the LST body:

$$\left. \begin{aligned} I_x \ddot{\phi} + \dot{\theta} \dot{\psi} (I_z - I_y) + I_{3z} v_3 \dot{\theta} - I_{2y} v_2 \dot{\psi} + I_{1x} \dot{v}_1 &= M_x \\ I_y \ddot{\theta} + \dot{\phi} \dot{\psi} (I_x - I_z) + I_{1x} v_1 \dot{\psi} - I_{3z} v_3 \dot{\phi} + I_{2y} \dot{v}_2 &= M_y \\ I_z \ddot{\psi} + \dot{\phi} \dot{\theta} (I_y - I_x) + I_{2y} v_2 \dot{\phi} - I_{1x} v_1 \dot{\theta} + I_{3z} \dot{v}_3 &= M_z \end{aligned} \right\} \quad (2.5)$$

(ii) Equations for the Reaction Wheel mounted parallel to x-axis:

$$\left. \begin{aligned} I_{1x} \ddot{\phi} + I_{1x} \dot{v}_1 &= M_{1x} \\ I_{1y} \ddot{\theta} + (I_{1x} - I_{1y}) \dot{\theta} \dot{\psi} + I_{1x} v_1 \dot{\psi} &= M_{1y} \\ I_{1y} \ddot{\psi} + (I_{1y} - I_{1x}) \dot{\theta} \dot{\phi} - I_{1x} v_1 \dot{\theta} &= M_{1z} \end{aligned} \right\} \quad (2.6)$$

(iii) Equations for the Reaction Wheel mounted parallel to y-axis

$$\left. \begin{aligned} I_{2z} \ddot{\phi} + (I_{2y} - I_{2z}) \dot{\phi} \dot{\psi} + I_{2y} v_2 \dot{\phi} &= M_{2x} \\ I_{2y} \ddot{\theta} + I_{2y} \dot{v}_2 &= M_{2y} \\ I_{2z} \ddot{\psi} + (I_{2z} - I_{2y}) \dot{\theta} \dot{\phi} - I_{2y} v_2 \dot{\psi} &= M_{2z} \end{aligned} \right\} \quad (2.7)$$

(iv) Equations for the Reaction Wheel mounted parallel to z-axis:

$$\left. \begin{aligned} I_{3x} \ddot{\theta} + (I_{3x} - I_{3z}) \dot{\phi} \dot{\theta} + I_{3z} v_3 \dot{\theta} &= M_{3x} \\ I_{3x} \ddot{\phi} + (I_{3z} - I_{3x}) \dot{\phi} \dot{\psi} + I_{3z} v_3 \dot{\phi} &= M_{3y} \\ I_{3z} \ddot{\psi} + I_{3z} \dot{v}_3 &= M_{3z} \end{aligned} \right\} \quad (2.8)$$



For further simplification, we will assume that the reaction wheels are small so that  $I_{1x} \ll I_x$ ,  $I_{2y} \ll I_y$ ,  $I_{3z} \ll I_z$  and they have one degree of freedom only. With these, equations (2.5)-(2.8) can be simplified into,

$$\left. \begin{aligned} I_x \ddot{\phi} + \dot{\theta} \dot{\psi} (I_z - I_y) + I_{1x} \dot{v}_1 &= M_x \\ I_y \ddot{\theta} + \dot{\phi} \dot{\psi} (I_x - I_z) + I_{2y} \dot{v}_2 &= M_y \\ I_z \ddot{\psi} + \dot{\phi} \dot{\theta} (I_y - I_x) + I_{3z} \dot{v}_3 &= M_z \end{aligned} \right\} \quad (2.9)$$

and

$$I_{1x} \ddot{\phi} + I_{1x} \dot{v}_1 = M_{1x} \quad (2.10)$$

$$I_{2y} \ddot{\theta} + I_{2y} \dot{v}_2 = M_{2y} \quad (2.11)$$

$$I_{3z} \ddot{\psi} + I_{3z} \dot{v}_3 = M_{3z} \quad (2.12)$$

Substitution of (2.10)-(2.12) into (2.9) will result in the following three equations describing the motions along the individual axes and their interconnections:

$$\left. \begin{aligned} I_x \ddot{\phi} + (I_z - I_y) \dot{\theta} \dot{\psi} &= (M_x - M_{1x}) \\ I_y \ddot{\theta} + (I_x - I_z) \dot{\phi} \dot{\psi} &= (M_y - M_{2y}) \\ I_z \ddot{\psi} + (I_y - I_x) \dot{\theta} \dot{\phi} &= (M_z - M_{3z}) \end{aligned} \right\} \quad (2.13)$$

It is now necessary to evaluate the various torques. Since the internal torques on the reaction wheels are small, it may be assumed that these are proportional to the control signals actuating the wheels. Hence,

$$\left. \begin{aligned} M_{1x} &= -K_1 u_1 \\ M_{2y} &= -K_2 u_2 \\ M_{3z} &= -K_3 u_3 \end{aligned} \right\} \quad (2.14)$$

Where  $K_1$ ,  $K_2$  and  $K_3$  are the drive motor constants (the negative signs in (2.14) merely indicate the directions of these torques).

The external torques acting on the body of the LST are mainly environmental disturbance forces and are composed of gravity-gradient, magnetic, aerodynamic and solar pressure torques. The latter two will be negligibly small compared to the others and will usually be accounted for in control system designs by considering them as equivalent zero-mean stationary white noise processes. The gravity-gradient and magnetic torques can be represented as purely deterministic signals involving a constant term and a sinusoidal function of time with twice orbital rate. Hence, following the analysis in [11] the external torques can be obtained as,

$$\left. \begin{aligned} M_x &= \{\gamma_{11} + \gamma_{12} \cos(\omega t + x) + s_1\} I_x \\ M_y &= \{\gamma_{21} + \gamma_{22} \cos(\omega t + x) + s_2\} I_y \\ M_z &= \{\gamma_{31} + \gamma_{32} \cos(\omega t + x) + s_3\} I_z \end{aligned} \right\} \quad (2.15)$$

Where  $\gamma_{ij}$ ,  $i = 1, 2, 3$ , are constants that can be determined [11] from the inertia components  $I_x$ ,  $I_y$ ,  $I_z$ , the magnitude of the LST dipole moment and the earth's magnetic field intensity; and  $s_i$ ,  $i = 1, 2, 3$ , are white-noise processes characterizing the aerodynamic and solar pressure torques.

Substitution of (2.14) and (2.15) in (2.13) and further simplification results in the following system of equations:

$$\left. \begin{aligned} \ddot{\phi} + \alpha_1 \dot{\theta} \dot{\psi} &= \beta_1 u_1 + M_x \\ \ddot{\theta} + \alpha_2 \dot{\phi} \dot{\psi} &= \beta_2 u_2 + M_y \\ \ddot{\psi} + \alpha_3 \dot{\phi} \dot{\theta} &= \beta_3 u_3 + M_z \end{aligned} \right\} \quad (2.16)$$

where  $\alpha_1 = \frac{(I_z - I_y)}{I_x}$ ,  $\alpha_2 = \frac{(I_x - I_z)}{I_y}$ ,  $\alpha_3 = \frac{(I_y - I_x)}{I_z}$ ,  $\beta_1 = \frac{K_1}{I_x}$ ,  $\beta_2 = \frac{K_2}{I_y}$ ,

$\beta_3 = \frac{K_3}{I_z}$  and  $M_x, M_y, M_z$  are the external disturbance torques given by (2.15).

It is now simple to obtain a state-space representation of the LST by choosing the state-vector

$$x = [\phi, \dot{\phi}, \theta, \dot{\theta}, \psi, \dot{\psi}]^T, \quad (2.17)$$

which results in the time-invariant model,

$$\dot{x} = Ax + Bu + h(x) + FM \quad (2.18)$$

where

$$A = \begin{bmatrix} 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & 0 & 0 & 0 \end{bmatrix}; \quad B = \begin{bmatrix} 0 & 0 & 0 \\ \beta_1 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & \beta_2 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & \beta_3 \end{bmatrix}$$

$$h(x) = \begin{bmatrix} 0 \\ -\alpha_1 \dot{\theta} \dot{\psi} \\ 0 \\ -\alpha_2 \dot{\phi} \dot{\psi} \\ 0 \\ -\alpha_3 \dot{\phi} \dot{\theta} \end{bmatrix} \quad \text{and} \quad F = \begin{bmatrix} 0 & 0 & 0 \\ 1 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

The diagonal structure of the matrices  $A$ ,  $B$  and  $F$  permits us to partition the state-vector as,

$$x = [x_1, x_2, x_3]^T \quad (2.19)$$

where

$$x_1 = \begin{bmatrix} x_{11} \\ x_{12} \end{bmatrix} = \begin{bmatrix} \phi \\ \dot{\phi} \end{bmatrix}, \quad x_2 = \begin{bmatrix} x_{21} \\ x_{22} \end{bmatrix} = \begin{bmatrix} \theta \\ \dot{\theta} \end{bmatrix}, \quad x_3 = \begin{bmatrix} x_{31} \\ x_{32} \end{bmatrix} = \begin{bmatrix} \psi \\ \dot{\psi} \end{bmatrix}.$$

With this, (2.18) can be described as a set of interconnected subsystems,

$$\dot{x}_i = A_i x_i + b_i u_i + h_i(x) + f_i d_i, \quad i = 1, 2, 3, \quad (2.20)$$

where

$$A_i = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix}, \quad b_i = \begin{bmatrix} 0 \\ \beta_i \end{bmatrix}, \quad f_i = \begin{bmatrix} 0 \\ 1 \end{bmatrix}, \quad i = 1, 2, 3,$$

and

$$h_1(x) = \begin{bmatrix} 0 \\ -\alpha_1 x_{22} x_{32} \end{bmatrix}, \quad h_2(x) = \begin{bmatrix} 0 \\ -\alpha_2 x_{32} x_{12} \end{bmatrix}, \quad h_3(x) = \begin{bmatrix} 0 \\ -\alpha_3 x_{12} x_{22} \end{bmatrix}$$

with  $d_1 = M_x$ ,  $d_2 = M_y$  and  $d_3 = M_z$  being the external disturbances.

It may be observed that when  $h_i(x) \equiv 0$ ,  $i = 1, 2, 3$ , (2.20) represents three decoupled subsystems which describe the motions of the spacecraft along the three axes. However,  $h_i(x)$  are not zero and constitute the interconnections among the subsystems, thus making an analysis based on the smaller-dimensional decoupled subsystems alone inaccurate.

The system represented by (2.20), is driven by the disturbance forces  $d_i$  in addition to the control signals  $u_i$ . However, these external disturbances can be completely cancelled by constructing a disturbance accommodating controller as described in [3, 1]. This involves the determination of a suitable differential equation model for the disturbances and with the augmentation of the disturbance variables with the state variables of the system, designing a feedback controller that counteracts the disturbance forces by feeding back the estimated disturbance variables. Although this analysis is conducted for a single-axis model of the LST (only for the pitch motion control) in [1], a straightforward extension that uses three separate disturbance accommodating controllers can be obtained for the three-axis model presently considered. Due to the above reason, we will ignore the disturbance terms from our model and conduct all fur-

ther analysis on the system,

$$\dot{x}_i = A_i x_i + b_i u_i + h_i(x) , \quad i = 1, 2, 3 , \quad (2.21)$$

obtained from (2.20) with the substitution  $d_i \equiv 0$  .

### 3. STABILIZATION

When a complex dynamic system is given as a number of locally controlled interconnected subsystems, it can be stabilized by a multilevel control scheme [6 - 8] based upon the decomposition-aggregation stability analysis [1 - 3]. In the scheme, the dimensionality problem is resolved by carrying out all operations on the subsystem level. Both local and global controllers can be involved. The local controllers are introduced to stabilize each decoupled subsystem by any of the classical techniques such as the pole-shifting by state feedback, root-locus, parameter plane method, etc. The global controllers minimize the effect of interactions among the subsystems. Finally, the aggregate system is constructed on the higher hierarchical level to conclude stability of the overall composite system.

It is important to note that the proposed stabilization produces large-scale systems which are connectively stable [1 - 3]. That is, stability is invariant under structural perturbations whereby subsystems are disconnected and connected again in various ways during the operation of the system. Furthermore, the stabilized systems have wide tolerance to nonlinearities in the interactions between the subsystems.

After we outline the multilevel control scheme for stabilization of large-scale systems, we will consider a class of dynamic systems which can be always stabilized by the scheme using local controllers only. Since the LST model developed in the preceding section falls in that class, we will be able to effectively design the feedback control which stabilizes the LST.

#### 3.1. *Multilevel Control*

Let us consider a linear dynamic system

$$\dot{x} = Ax + Bu, \quad (3.1)$$

where  $x(t) \in R^n$  is the state of the system,  $u(t) \in R^s$  is the input to the system, and  $A$  and  $B$  are constant  $n \times n$  and  $n \times s$  matrices. We assume that the system is brought into the input-decentralized form

$$\dot{x}_i = A_i x_i + \sum_{\substack{j=1 \\ j \neq i}}^s A_{ij} x_j + b_i u_i, \quad i = 1, 2, \dots, s \quad (3.2)$$

where  $x_i(t) \in R^{n_i}$  is the state of the  $i$ -th subsystem, and  $u_i(t) \in R$  is the corresponding local control, so that

$$R^n = R^{n_1} \times R^{n_2} \times \dots \times R^{n_s}, \quad (3.3)$$

and each pair  $(A_i, b_i)$  is controllable.

In (3.2), the matrices  $A_i, A_{ij}$ , and the vectors  $b_i$  have appropriate dimensions. As shown in reference [ 6 ], any linear dynamic system (3.1) can be represented by its input-decentralized form (3.2).

To stabilize the system (3.2), we apply the decentralized feedback control

$$u_i(t) = u_i^l(t) + u_i^g(t) \quad (3.4)$$

where  $u_i(t)$  is chosen as a local control law

$$u_i^l = -k_i^T x_i, \quad (3.5)$$

with a constant vector  $k_i \in R^{n_i}$ , and  $u_i^g(t)$  is chosen as a global control law

$$u_i^g = - \sum_{\substack{j=1 \\ j \neq i}}^s k_{ij}^T x_j, \quad (3.6)$$

where  $k_{ij} \in R^{n_j}$  are constant vectors.

By substituting the control (3.4) into (3.2), we get the closed-loop sys-

tem as

$$\dot{x}_i = (A_i - b_i k_i^T) x_i + \sum_{\substack{j=1 \\ j \neq i}}^s (A_{ij} - b_i k_{ij}^T) x_j, \quad i = 1, 2, \dots, s. \quad (3.7)$$

Since each pair  $(A_i, b_i)$  is controllable, a simple choice of  $k_i$  can be always made [13] to place the eigenvalues of  $A_i - B_i k_i^T$  at any desired locations  $-\sigma_1^i \pm j\omega_1^i, \dots, \sigma_p^i \pm j\omega_p^i, -\sigma_{p+1}^i, \dots, -\sigma_{n_i}^i$  ( $\sigma_q^i > 0$ ;  $q = 1, 2, \dots, n_i$ , and  $1 \leq p \leq n_i$ ). Then, each uncoupled subsystem

$$\dot{x}_i = (A_i - b_i k_i^T) x_i, \quad i = 1, 2, \dots, s \quad (3.8)$$

is stabilized with a degree of exponential stability

$$\pi_i = \min_q \sigma_q^i. \quad (3.9)$$

To provide a Liapunov function [5-8] with the exact estimate of  $\pi_i$  for each decoupled subsystem, we apply to (3.8) the linear nonsingular transformation

$$x_i = T_i \tilde{x}_i, \quad (3.10)$$

to get the system (3.8) as

$$\dot{\tilde{x}}_i = \Lambda_i \tilde{x}_i, \quad (3.11)$$

where  $\Lambda_i = T_i^{-1} (A_i - b_i k_i^T) T_i$  has the quasidiagonal form

$$\Lambda_i = \text{diag} \left\{ \begin{bmatrix} -\sigma_1^i & \omega_1^i \\ \omega_1^i & -\sigma_1^i \end{bmatrix}, \dots, \begin{bmatrix} -\sigma_p^i & \omega_p^i \\ \omega_p^i & -\sigma_p^i \end{bmatrix}, -\sigma_{p+1}^i, \dots, -\sigma_{n_i}^i \right\}. \quad (3.12)$$



For the system (3.11), we choose the Liapunov function  $v_i: R^{n_i} \rightarrow R_+$ ,

$$v_i(\tilde{x}_i) = (\tilde{x}_i^T \tilde{H}_i \tilde{x}_i)^{1/2}, \quad (3.13)$$

where

$$\Lambda_i^T \tilde{H}_i + \tilde{H}_i \Lambda_i = -\tilde{G}_i, \quad (3.14)$$

and

$$\tilde{G}_i = 2\theta_i \text{diag} \{ \sigma_1^i, \sigma_1^i, \dots, \sigma_p^i, \sigma_p^i, \sigma_{p+1}^i, \dots, \sigma_{n_i}^i \}, \quad \tilde{H}_i = \theta_i I_i. \quad (3.15)$$

In (3.15),  $\theta_i > 0$  is an arbitrary constant and  $I_i$  is the  $n_i \times n_i$  identity matrix.

The aggregate comparison system involving the vector Liapunov function  $v: R^n \rightarrow R_+$ ,

$$v = (v_1, v_2, \dots, v_s)^T, \quad (3.16)$$

is obtained for the transformed system (3.7),

$$\dot{\tilde{x}}_i = \Lambda_i \tilde{x}_i + \sum_{\substack{j=1 \\ j \neq i}}^s (\tilde{A}_{ij} - \tilde{b}_i \tilde{k}_{ij}^T) \tilde{x}_j, \quad i = 1, 2, \dots, s \quad (3.17)$$

where  $\tilde{A}_{ij} = T_i^{-1} A_{ij} T_j$ ,  $\tilde{b}_i = T_i^{-1} b_i$ ,  $\tilde{k}_{ij}^T = k_{ij}^T T_j$ , and using the Liapunov functions  $v_i(\tilde{x}_i)$  defined in (3.13). Using the aggregation method presented in [1 - 5], we construct the comparison system

$$\dot{v} \leq \tilde{W}v, \quad (3.18)$$

where the constant  $s \times s$  matrix  $\tilde{W} = (\tilde{w}_{ij})$  has the elements defined as

$$\tilde{w}_{ij} = -\delta_{ij} \pi_i + (1 - \delta_{ij}) \tilde{\xi}_{ij}, \quad (3.19)$$

where  $\delta_{ij}$  is the Kronecker symbol,  $\pi_i$  is defined in (3.9), and

$$\tilde{\xi}_{ij} = \lambda_M^{1/2} [(\tilde{A}_{ij} - \tilde{b}_i \tilde{k}_{ij}^T)^T (\tilde{A}_{ij} - \tilde{b}_i \tilde{k}_{ij}^T)] \quad (3.20)$$

where  $\lambda_M$  is the maximum eigenvalue of the indicated matrix.

As known [1 - 3], global asymptotic stability of the system (3.17) and, therefore, original system (3.2), is implied by the Sevastyanov-Kotelyanski conditions [14], which for  $\tilde{W} = (\tilde{w}_{ij})$  defined by (3.19) and (3.20) have the following form

$$(-1)^k \begin{vmatrix} -\pi_1 & \tilde{\xi}_{12} & \cdots & \tilde{\xi}_{1k} \\ \tilde{\xi}_{21} & -\pi_2 & \cdots & \tilde{\xi}_{2k} \\ \cdots & \cdots & \cdots & \cdots \\ \tilde{\xi}_{k1} & \tilde{\xi}_{k2} & \cdots & -\pi_k \end{vmatrix} > 0, \quad k = 1, 2, \dots, s. \quad (3.21)$$

To satisfy conditions (3.21), we choose the vectors  $\tilde{k}_{ij}$  in (3.20) so as to minimize the nonnegative numbers  $\tilde{\xi}_{ij}$  which reflect the strength of interconnections among the subsystems in (3.17). Such choice is provided by

$$\tilde{k}_{ij}^* = [(\tilde{b}_i^T \tilde{b}_i)^{-1} \tilde{b}_i^T \tilde{A}_{ij}]^T, \quad (3.22)$$

where  $(\tilde{b}_i^T \tilde{b}_i)^{-1} \tilde{b}_i^T$  is the Moore-Penrose generalized inverse of  $\tilde{b}_i$  [15].

The choice of  $\tilde{k}_{ij}^*$  in (3.22) produces the optimal aggregate matrix  $\tilde{W}^*$  in the sense that  $\tilde{W}^* \leq \tilde{W}$  (that is,  $\tilde{W}^* - \tilde{W} \leq 0$ ) is valid for all  $\tilde{k}_{ij}$ . That is equivalent to saying [16] that  $\lambda_M(\tilde{W}^*) \leq \lambda_M(\tilde{W})$  for all  $\tilde{k}_{ij}$ . Since conditions (3.21) are necessary and sufficient for  $\lambda_M(\tilde{W}) < 0$ , that is, for stability of  $\tilde{W}$ , the choice  $\tilde{k}_{ij} = \tilde{k}_{ij}^*$  is justified.

To conclude stability of the overall system (3.17) with the optimal choice  $\tilde{k}_{ij} = \tilde{k}_{ij}^*$ , which is

$$\dot{\tilde{x}}_i = \Lambda_i \tilde{x}_i + [I_i - \tilde{b}_i (\tilde{b}_i^T \tilde{b}_i)^{-1} \tilde{b}_i^T] \sum_{\substack{j=1 \\ j \neq i}}^s \tilde{A}_{ij} \tilde{x}_j, \quad j = 1, 2, \dots, s \quad (3.23)$$

we apply the determinantal inequalities (3.21) to the optimal aggregate matrix  $\tilde{W}^* = (\tilde{w}_{ij}^*)$  defined by (3.19) and  $\tilde{\xi}_{ij} = \tilde{\xi}_{ij}^* = \lambda_M^{1/2} \{ \tilde{A}_{ij}^T [I_i - \tilde{b}_i (\tilde{b}_i^T \tilde{b}_i)^{-1} \tilde{b}_i^T] \tilde{A}_{ij} \}$ .

We arrive at the following:

Theorem 3.1. *The linear control system (3.2) is stabilized by the linear control laws*

$$u_i = -k_i^T x_i - \sum_{\substack{j=1 \\ j \neq i}}^s k_{ij}^T x_j, \quad i = 1, 2, \dots, s \quad (3.24)$$

where  $k_{ij}^T = \tilde{k}_{ij}^{*T} \Gamma_j^{-1}$ , if the corresponding  $s \times s$  aggregate matrix

$$\tilde{W}^* = [-\delta_{ij} \pi_i + (1 - \delta_{ij}) \tilde{\xi}_{ij}^*] \quad (3.25)$$

satisfies conditions (3.21).

Successful application of the above theorem depends on appropriate choice of the eigenvalues for the decoupled subsystems (3.8). Once the subsystem eigenvalues are prescribed, the control law (3.24) and, thus, the gain vectors  $k_i$ ,  $k_{ij}$  in (3.24), are computed uniquely using the proposed algorithm. Therefore, if for computed gains  $k_i$ ,  $k_{ij}$ , the conditions (3.21) are not met, a reassignment of the subsystems eigenvalues is required. The search for an appropriate set of subsystems eigenvalues can be aided by the interactive computer program described in Section A.1. The efficiency of the computer program relies on the low order of the subsystems and the simplicity in testing the Sevastyanov-Kotelyanskii conditions (3.21). Furthermore, the computerized procedure provides a considerable freedom to the designer to apply his understanding of the system and the familiarity with the method to come up with a successful design.

### 3.2 An Illustrative Example

Let us consider a system (3.1) described by the equation

$$\dot{x} = \begin{bmatrix} 1 & 11.50 & 86.50 & 4 & 22.50 \\ 0.45 & 0 & -4.09 & 8.91 & -0.82 \\ 0.18 & 1 & 8.36 & 0.36 & 3.27 \\ \hline 0 & 1.25 & 14.75 & 5 & 2.75 \\ 0.18 & 0 & -9.82 & 0.18 & -6.36 \end{bmatrix} x + \begin{bmatrix} 1 & 0 \\ 1 & 0 \\ 0 & 0 \\ \hline 0 & 1 \\ 0 & 1 \end{bmatrix} u. \quad (3.26)$$

The eigenvalues of the matrix  $A$  corresponding to (3.26) are

$$\lambda_{1,2} = 0.76 \pm j 1.83, \lambda_3 = 11.54, \lambda_4 = -3.89, \lambda_5 = -1.15, \quad (3.27)$$

and the system is unstable.

To stabilize the system (3.26), we start with its input-decentralized representation (3.2) given as

$$\begin{aligned} \dot{x}_1 &= \begin{bmatrix} 1 & 11.50 & 86.50 \\ 0.45 & 0 & -4.09 \\ 0.18 & 1 & 8.36 \end{bmatrix} x_1 + \begin{bmatrix} 4 & 22.50 \\ 8.91 & -0.82 \\ 0.36 & 3.27 \end{bmatrix} x_2 + \begin{bmatrix} 1 \\ 1 \\ 0 \end{bmatrix} u_1 \\ \dot{x}_2 &= \begin{bmatrix} 5 & 2.75 \\ 0.18 & -6.36 \end{bmatrix} x_2 + \begin{bmatrix} 0 & 1.25 & 14.75 \\ 0.18 & 0 & -9.82 \end{bmatrix} x_1 + \begin{bmatrix} 1 \\ 1 \end{bmatrix} u_2. \end{aligned} \quad (3.28)$$

and transform each subsystem into its comparison form [13] to get

$$\begin{aligned} \dot{x}_1^c &= \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -8.86 & 8.50 & 9.36 \end{bmatrix} x_1^c + \begin{bmatrix} 3.20 & 1.98 \\ -14.72 & 0.49 \\ -7.92 & 36.01 \end{bmatrix} x_2^c + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} u_1 \\ \dot{x}_2^c &= \begin{bmatrix} 0 & 1 \\ 32.32 & -1.36 \end{bmatrix} x_2^c + \begin{bmatrix} 1.69 & 1.26 & 0.08 \\ -7.52 & -5.23 & 0.49 \end{bmatrix} x_1^c + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u_2 \end{aligned} \quad (3.29)$$

The transformation into the comparison form is of no conceptual significance, and is performed on the subsystem level for two practical reasons. First, it is convenient for subsystem stabilization by pole-assignment applying the state feedback and, secondly, the diagonal form (3.17) with no complex roots, can be obtained from the companion form (3.29) using the Vandermonde matrix  $T_i$  in (3.10) where  $x_i$  is replaced by  $x_i^C$ .

Now, by using the local feedback law (3.5) and vectors

$$\begin{aligned} k_1^T &= (1791.14, 458.50, 46.36) \\ k_2^T &= (33.82, 1.14) , \end{aligned} \quad (3.30)$$

we allocate the eigenvalues of the uncoupled subsystems (3.27) from

$$\begin{aligned} \lambda_1^1 &= 0.63 , & \lambda_1^2 &= 5.04 \\ \lambda_2^1 &= -1.39 , & \lambda_2^2 &= -6.41 \\ \lambda_3^1 &= 10.12 , & & \end{aligned} \quad (3.31)$$

to

$$\begin{aligned} \lambda_1^1 &= -10 , & \lambda_1^2 &= -1 \\ \lambda_2^1 &= -12 , & \lambda_2^2 &= -1.5 \\ \lambda_3^1 &= -15 . & & \end{aligned} \quad (3.32)$$

After the local stabilization, the interconnected subsystems have the quasi-diagonal form

$$\begin{aligned} \dot{\tilde{x}}_1 &= \begin{bmatrix} -10 & 0 & 0 \\ 0 & -12 & 0 \\ 0 & 0 & -15 \end{bmatrix} \tilde{x}_1 + \begin{bmatrix} -23.52 & -43.78 \\ 40.23 & 68.97 \\ -15.49 & -24.96 \end{bmatrix} \tilde{x}_2 + \begin{bmatrix} 0.1 \\ -0.17 \\ 0.07 \end{bmatrix} u_1^g \\ \dot{\tilde{x}}_2 &= \begin{bmatrix} -1 & 0 \\ 0 & -1.5 \end{bmatrix} \tilde{x}_2 + \begin{bmatrix} 179.95 & 247.53 & 367.40 \\ -182.58 & -249.03 & -365.95 \end{bmatrix} \tilde{x}_1 + \begin{bmatrix} 2 \\ -2 \end{bmatrix} u_2^g , \end{aligned} \quad (3.33)$$

which is not identical to (3.17). For the moment, we did not make use of the global control  $u_1^g, u_2^g$  in (3.33). In order to demonstrate the effect of the global controllers, we set  $\tilde{k}_{12} = \tilde{k}_{21} = 0$ .

From (3.9) and (3.32), we have  $\pi_1 = 10$ ,  $\pi_2 = 1$ . Using (3.20) and (3.33) we compute  $\tilde{\xi}_{12} = 98.51$ ,  $\tilde{\xi}_{21} = 676.68$ . The aggregate matrix  $\tilde{W}$  in (3.18) is obtained as

$$\tilde{W} = \begin{bmatrix} -10 & 98.51 \\ 676.68 & -1 \end{bmatrix}, \quad (3.34)$$

which does not satisfy the conditions (3.21). Therefore, we cannot conclude stability of the overall system.

Let us use now the global control specified by (3.22),

$$\begin{aligned} \tilde{k}_{12}^{*T} &= (-238.95, -415.34) \\ \tilde{k}_{21}^{*T} &= (90.63, 124.14, 183.33) \end{aligned} \quad (3.35)$$

which yields the subsystems (3.33) as

$$\begin{aligned} \dot{\tilde{x}}_1 &= \begin{bmatrix} -10 & 0 & 0 \\ 0 & -12 & 0 \\ 0 & 0 & -15 \end{bmatrix} \tilde{x}_1 + \begin{bmatrix} 0.37 & -2.25 \\ 0.40 & -0.25 \\ 0.44 & 2.73 \end{bmatrix} \tilde{x}_2 \\ \dot{\tilde{x}}_2 &= \begin{bmatrix} -1 & 0 \\ 0 & -1.5 \end{bmatrix} \tilde{x}_2 + \begin{bmatrix} -1.31 & -0.75 & 0.72 \\ -1.31 & -0.75 & 0.72 \end{bmatrix} \tilde{x}_1, \end{aligned} \quad (3.36)$$

and the aggregate matrix

$$\tilde{W}^* = \begin{bmatrix} -10 & 3.55 \\ 2.37 & -1 \end{bmatrix}, \quad (3.37)$$

which satisfies the conditions (3.21). Therefore, by theorem 1 the system (3.28) is stabilized by the control law (3.24) determined by the gains (3.30)

and (3.35). The eigenvalues of the overall closed-loop system

$$\dot{\tilde{x}} = \begin{bmatrix} -10 & 0 & 0 & 0.37 & -2.25 \\ 0 & -12 & 0 & 0.40 & -0.25 \\ 0 & 0 & -15 & 0.44 & 2.73 \\ \hline -1.31 & -0.75 & 0.72 & -1 & 0 \\ -1.31 & -0.75 & 0.72 & 0 & -1.5 \end{bmatrix} \tilde{x} \quad (3.38)$$

corresponding to (3.36), are

$$\lambda_{1,2} = -1.03 \pm j0.16, \lambda_3 = -10.27, \lambda_4 = -11.99, \lambda_5 = -15.17, \quad (3.39)$$

which have negative real parts.

It is also interesting to note that an upper estimate of the degree  $\pi$  of exponential stability of the system (3.1) is provided by the aggregate matrix  $\tilde{W}^*$  since, in general  $\pi \leq \min_i \pi_i$ . In other words, the degree of exponential stability of the overall system  $\pi$  stabilized by the proposed method, is smaller than the degree of exponential stability of each decoupled subsystem.

### 3.3 Local Stabilization

In this section, we consider a class of linear input-decentralized large-scale systems which can always be stabilized by only local feedback control applied around each subsystem. This class of systems is characterized by the comparison form of the subsystem matrices and the lower diagonal form of the interconnection matrices.

Let us consider again the system

$$\dot{x}_i = A_i x_i + \sum_{\substack{j=1 \\ j \neq i}}^s A_{ij} x_j + b_i u_i, \quad i = 1, 2, \dots, s \quad (3.2)$$

where the  $n_i \times n_i$  matrix  $A_i$  and the  $n_i$  vector  $b_i$  are

$$A_i = \begin{bmatrix} 0 & 1 & \dots & 0 \\ 0 & 0 & \dots & 0 \\ \dots & \dots & \dots & \dots \\ 0 & 0 & \dots & 1 \\ -a_1^i & -a_2^i & \dots & -a_{n_i}^i \end{bmatrix}, \quad b_i = \begin{bmatrix} 0 \\ 0 \\ \vdots \\ 0 \\ 1 \end{bmatrix} \quad (3.40)$$

and the  $n_i \times n_j$  matrices  $A_{ij} = (a_{pq}^{ij})$  are such that

$$a_{pq}^{ij} = 0, \quad p < q \quad (3.41)$$

where  $p = 1, 2, \dots, n_i$  and  $q = 1, 2, \dots, n_j$ .

In order to stabilize system (3.2) characterized by (3.40) and (3.41), we apply the local control

$$u_i^{\ell} = -k_i^T x_i, \quad (3.5)$$

and get (3.2) as

$$\dot{x}_i = (A_i - b_i k_i^T) x_i + \sum_{\substack{j=1 \\ j \neq i}}^s A_{ij} x_j, \quad i = 1, 2, \dots, s. \quad (3.42)$$

Gain vectors  $k_i$  are chosen so that each matrix  $A_i - b_i k_i^T$  has a set  $L_i$  of distinct real eigenvalues  $\lambda_p^i$  defined by

$$L_i = \{\lambda_p^i: \lambda_p^i = -\alpha \sigma_p^i; \alpha \geq 1, \sigma_p^i > 0, p = 1, 2, \dots, n_i\} \\ i = 1, 2, \dots, s. \quad (3.43)$$

The positive constant  $\alpha$  is to be determined, so that the overall system (3.2) is stabilized.

Following the development in Section 3.1, we transform (3.42) into

$$\dot{\tilde{x}}_i = \Lambda_i \tilde{x}_i + \sum_{\substack{j=1 \\ j \neq i}}^s \tilde{A}_{ij} \tilde{x}_j, \quad i = 1, 2, \dots, s \quad (3.44)$$



where the transformation (3.10) is used to get

$$\Lambda_i = T_i^{-1} (A_i - b_i k_i^T) T_i, \quad \tilde{A}_{ij} = T_i^{-1} A_{ij} T_j. \quad (3.45)$$

with  $\Lambda_i$  in the quasisdiagonal form

$$\Lambda_i = \text{diag} \{-\alpha_1^i, -\alpha_2^i, \dots, -\alpha_{n_i}^i\}. \quad (3.46)$$

In this case, the transformation matrix  $T_i$  can be factorized as

$$T_i = R_i \hat{T}_i, \quad (3.47)$$

where

$$R_i = \text{diag} \{1, \alpha, \dots, \alpha^{n_i-1}\}, \quad (3.48)$$

and  $\hat{T}_i$  is the Vandermonde matrix

$$\hat{T}_i = \begin{bmatrix} 1 & 1 & \dots & 1 \\ -\sigma_1^i & -\sigma_2^i & \dots & -\sigma_{n_i}^i \\ \vdots & \vdots & & \vdots \\ (-\sigma_1^i)^{n_i-1} & (-\sigma_2^i)^{n_i-1} & \dots & (-\sigma_{n_i}^i)^{n_i-1} \end{bmatrix} \quad (3.49)$$

For the moment, we consider the free uncoupled subsystems

$$\dot{\tilde{x}}_i = \Lambda_i \tilde{x}_i, \quad i = 1, 2, \dots, s. \quad (3.50)$$

Each subsystem (3.50) is stabilized with a degree of exponential stability

$$\pi_i = \alpha \hat{\pi}_i \quad (3.51)$$

where

$$\hat{\pi}_i = \min_p \sigma_p^i. \quad (3.52)$$

Now, we choose again the Liapunov function  $v: R^{n_i} \rightarrow R_+$ ,

$$v_i(\tilde{x}_i) = (\tilde{x}_i^T \tilde{H}_i \tilde{x}_i)^{1/2} \quad (3.13)$$

where

$$\Lambda_i^T \tilde{H}_i + \tilde{H}_i \Lambda_i = -\tilde{G}_i \quad (3.14)$$

and

$$\tilde{G}_i = 2\theta_i \text{diag} \{ \alpha \sigma_1^i, \alpha \sigma_2^i, \dots, \alpha \sigma_{n_i}^i \}, \quad \tilde{H}_i = \theta_i I_i. \quad (3.53)$$

The aggregate system

$$\dot{v} \leq \tilde{W}v \quad (3.18)$$

is formed as in Section 3.1 computing the elements  $\tilde{w}_{ij}$  of the aggregate matrix  $\tilde{W}$  with

$$\tilde{\xi}_{ij} = \lambda_M^{1/2} (\tilde{A}_{ij}^T \tilde{A}_{ij}) \quad (3.54)$$

and

$$\tilde{A}_{ij} = \hat{T}_i^{-1} R_i^{-1} A_{ij} R_j \hat{T}_j. \quad (3.55)$$

Our ability to stabilize the system depends ultimately on satisfying the Sevastyanov-Kotelyanskii conditions (3.21) by the aggregate matrix  $\tilde{W} = (\tilde{w}_{ij})$  defined by

$$\tilde{w}_{ij} = -\delta_{ij} v_i + (1 - \delta_{ij}) \tilde{\xi}_{ij}. \quad (3.19)$$

Since the matrix  $\tilde{W}$  has nonnegative off-diagonal elements, it is a well-known fact [ 16 ] that the conditions (3.21) are equivalent to the quasidominant diagonal property of  $\tilde{W}$ ,

$$d_j |\tilde{w}_{jj}| > \sum_{\substack{i=1 \\ i \neq j}}^s d_i |\tilde{w}_{ij}|, \quad j = 1, 2, \dots, s \quad (3.56)$$

where  $d_i$ 's are positive numbers. Apparently, we can make the matrix  $\tilde{W}$  sat-

isfy conditions (3.20), if we can increase the diagonal elements  $\tilde{w}_{ii}$  sufficiently large while keeping the off-diagonal elements  $\tilde{w}_{ij}$  bounded. This is exactly the case with the class of systems under consideration. We notice that the diagonal elements ( $i = j$ ),

$$\tilde{w}_{ii} = -\alpha \hat{n}_i, \quad (3.57)$$

depend linearly on the adjustable parameter  $\alpha$ . The off-diagonal elements ( $i \neq j$ ),

$$\tilde{w}_{ij} = \tilde{\xi}_{ij}(\alpha), \quad (3.58)$$

are bounded functions of  $\alpha$ . To see this, we note that the elements  $\alpha^{q-p} a_{pq}^{ij}$  of the matrices  $R_i^{-1} A_{ij} R_j$  are either zero for  $p < q$  due to (3.41), or they are bounded for  $p \geq q$  due to nonpositive powers of  $\alpha$ . We have

$$\lim_{\alpha \rightarrow +\infty} R_i^{-1} A_{ij} R_j = D_{ij}, \quad (3.59)$$

where the matrix  $D_{ij} = (d_{pq}^{ij})$  is defined by:  $d_{pq}^{ij} = a_{pq}^{ij}$ , when  $p = q$ , and  $d_{pq}^{ij} = 0$ , when  $p \neq q$ . From (3.55) and (3.59), we define  $\tilde{D}_{ij} = \hat{T}_i^{-1} D_{ij} \hat{T}_j$  and conclude from

$$\lim_{\alpha \rightarrow +\infty} \tilde{\xi}_{ij}(\alpha) = \lambda_M^{\frac{1}{2}}(\tilde{D}_{ij}^T \tilde{D}_{ij}), \quad (3.60)$$

that the off-diagonal elements  $\tilde{w}_{ij}$  are bounded in  $\alpha$ .

Therefore, for the selected class of dynamic systems we can always choose a sufficiently large parameter  $\alpha$ , and use local linear feedback control to stabilize the systems. From (3.43), we see that by increasing the value of  $\alpha$ , we move the subsystem eigenvalues away from the origin, thus, increasing the degree of exponential stability of each subsystem. This, however, requires an increase of the local feedback gains in the course of stabilization.

### 3.4. An Illustrative Example

Let us illustrate the local stabilization procedure using the following example:

$$\dot{x} = \begin{bmatrix} 0 & 1 & 0 & | & 2 & 0 \\ 0 & 0 & 1 & | & 3 & 4 \\ -2 & -1 & -1 & | & 2 & 1 \\ \hline 4 & 0 & 0 & | & 0 & 1 \\ 5 & 6 & 0 & | & -3 & -2 \end{bmatrix} x + \begin{bmatrix} 0 & 0 \\ 0 & 0 \\ 1 & 0 \\ \hline 0 & 0 \\ 0 & 1 \end{bmatrix} u. \quad (3.61)$$

The eigenvalues of the system matrix  $A$  corresponding to (3.61), are

$$\lambda_1 = 1.7244, \lambda_2 = 5.1042, \lambda_3 = -1.2633, \lambda_{4,5} = -4.2826 \pm j1.7755 \quad (3.62)$$

and the system (3.61) is unstable.

The system (3.61) can be decomposed as

$$\dot{x}_1 = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -2 & -1 & -1 \end{bmatrix} x_1 + \begin{bmatrix} 2 & 0 \\ 3 & 4 \\ 2 & 1 \end{bmatrix} x_2 + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} u_1 \quad (3.63a)$$

$$\dot{x}_2 = \begin{bmatrix} 0 & 1 \\ -3 & -2 \end{bmatrix} x_2 + \begin{bmatrix} 4 & 0 & 0 \\ 5 & 6 & 0 \end{bmatrix} x_1 + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u_2 \quad (3.63b)$$

The eigenvalues of the subsystem (3.63a) are moved from

$$\lambda_1^1 = -1.3532, \lambda_{2,3}^1 = 0.1766 \pm j1.2028 \quad (3.64)$$

to the new locations

$$\lambda_1^1 = -\sigma_1^1 = -1, \lambda_2^1 = -\sigma_2^1 = -2, \lambda_3^1 = -\sigma_3^1 = -3 \quad (3.65)$$

applying the local control (3.5) and

$$k_1^T = (4, 10, 5). \quad (3.66)$$

Similarly, the eigenvalues of the subsystem (3.63b) are changed from

$$\lambda_{1,2}^2 = -1 + j1.4142, \quad (3.67)$$

to

$$\lambda_1^2 = -\sigma_1^2 = -1, \quad \lambda_2^2 = -\sigma_2^2 = -2 \quad (3.68)$$

applying the local control (3.5) and

$$k_2^T = (-1, 1). \quad (3.69)$$

Referring to (3.46), we see that in (3.65) and (3.68), the parameter  $\alpha = 1$ .

We construct the transformation matrices  $R_1, R_2, \hat{T}_1, \hat{T}_2$  for  $\alpha > 1$  as

$$R_1 = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \alpha & 0 \\ 0 & 0 & \alpha^2 \end{bmatrix}, \quad \hat{T}_1 = \begin{bmatrix} 1 & 1 & 1 \\ -1 & -2 & -3 \\ 1 & 4 & 9 \end{bmatrix}$$

$$R_2 = \begin{bmatrix} 1 & 0 \\ 0 & \alpha \end{bmatrix}, \quad \hat{T}_2 = \begin{bmatrix} 1 & 1 \\ -1 & -2 \end{bmatrix}. \quad (3.70)$$

The numbers  $\pi_1, \pi_2$  are both set to one. Then, the aggregation matrix of (3.18) defined by (3.57) is given as

$$\tilde{W} = \begin{bmatrix} -\alpha & \tilde{\xi}_{12} \\ \tilde{\xi}_{21} & -\alpha \end{bmatrix}, \quad (3.71)$$

which for  $\alpha = 1$  takes the form

$$\tilde{W} = \begin{bmatrix} -1 & 17.0011 \\ 12.2936 & -1 \end{bmatrix} \quad (3.72)$$

where

$$\tilde{\xi}_{12} = \lambda_M^{\frac{1}{2}} (\hat{T}_1^{-1} A_{12} \hat{T}_2), \quad \tilde{\xi}_{21} = \lambda_M^{\frac{1}{2}} (\hat{T}_2^{-1} A_{21} \hat{T}_1)$$

and  $A_{12}, A_{21}$  are specified in (3.63).

It is obvious that the matrix  $\tilde{W}$  in (3.72) does not satisfy the inequalities (3.21).

From (3.63) and (3.59), we find that

$$D_{12} = \begin{bmatrix} 2 & 0 \\ 0 & 4 \\ 0 & 0 \end{bmatrix}, \quad D_{21} = \begin{bmatrix} 4 & 0 & 0 \\ 0 & 6 & 0 \end{bmatrix} \quad (3.73)$$

and for  $\alpha > 15$ , we have  $\tilde{\xi}_{12} \approx 32.55$ ,  $\tilde{\xi}_{21} \approx 18.98$ . Thus, for  $\alpha = 25$ , we have the aggregate matrix

$$\tilde{W} = \begin{bmatrix} -25 & 32.55 \\ 18.98 & -25 \end{bmatrix}, \quad (3.75)$$

which satisfies the conditions (3.21), and the overall system is stable. The corresponding eigenvalues of the overall closed-loop system are

$$\lambda_1 = -36.0364, \quad \lambda_{2,3} = -25.9599 \pm j3.5219, \quad \lambda_{4,5} = -68.5213 \pm j6.0474. \quad (3.76)$$

For the chosen value of  $\alpha = 25$ , we have the eigenvalue sets  $L_1$  and  $L_2$  defined in (3.43) given as

$$\begin{aligned} L_1 &= \{-\alpha\sigma_1^1, -\alpha\sigma_2^1, -\alpha\sigma_3^1\} = \{-25, -50, -75\} \\ L_2 &= \{-\alpha\sigma_1^2, -\alpha\sigma_2^2\} = \{-25, -50\}. \end{aligned} \quad (3.77)$$

The locations of the subsystem eigenvalues specified by  $L_1, L_2$  of (3.77), are achieved by the local state-variable feedback defined by (3.5) and

$$\begin{aligned} k_1^T &= (93748, 6874, 149) \\ k_2^T &= (1247, 73) \end{aligned} \quad (3.78)$$

The gains in (3.78) are relatively high which is due to the use of local controllers only. The gains can be considerably reduced by applying global controllers in the multilevel scheme outlined in Section 3.1 and illustrated in Section 3.2.

### 3.5 Application to LST

In this section, we design a control system for the nonlinear model of LST described in Section 2, by using only the local linear controllers as proposed in Section 3.3. This necessitates an application of the results obtained by Weissenberger [17] which are concerned with the finite regions of stability of large-scale systems rather than their global stability properties.

We notice that the LST model (2.20) belongs to a general class of systems described by the equations

$$\dot{x}_i = A_i x_i + a_i x_\ell^T \sum_{\substack{j=1 \\ j \neq i, \ell}}^s A_{ij} x_j + b_i u_i$$

$$i = 1, 2, \dots, s; \ell = \begin{cases} 1, & i = s \\ i+1, & i \neq s \end{cases} \quad (3.79)$$

where  $A_i$  are constant  $n_i \times n_i$  matrices,  $A_{ij}$  are  $n_\ell \times n_j$  constant matrices,  $a_i$  and  $b_i$  are  $n_i$  constant vectors.

To stabilize the system (3.79), we choose the local control

$$u_i = -k_i^T x_i, \quad i = 1, 2, \dots, s \quad (3.80)$$

so that each uncoupled subsystem

$$\dot{x}_i = (A_i - b_i k_i^T) x_i, \quad i = 1, 2, \dots, s \quad (3.81)$$

has a prescribed set of distinct eigenvalues

$$L_i = \{-\sigma_1^i \pm j\omega_1^i, \dots, \sigma_p^i \pm j\omega_p^i, -\sigma_{p+1}^i, \dots, -\sigma_{n_i}^i;$$

$$\sigma_q^i > 0; p, q = 1, 2, \dots, n_i\}, \quad i = 1, 2, \dots, s. \quad (3.82)$$

By using the transformation (3.10), the closed-loop system corresponding



to (3.79) is obtained as

$$\dot{\tilde{x}}_i = \Lambda_i \tilde{x}_i + \tilde{a}_i \tilde{x}_\ell^T \sum_{\substack{j=1 \\ j \neq i, \ell}}^s \tilde{A}_{ij} \tilde{x}_j, \quad i = 1, 2, \dots, s, \ell = \begin{cases} 1, & i = s \\ i+1, & i \neq s, \end{cases} \quad (3.83)$$

where  $\Lambda_i = T_i^{-1}(A_i - b_i k_i^T)T_i$  has the quasidiagonal form (3.12),  $\tilde{A}_{ij} = T_\ell^T A_{ij} T_j$ , and  $\tilde{a}_i = T_i^{-1} a_i$ .

We define the interaction function  $h: T \times R^n \rightarrow R^{n_i}$  among the subsystems of (3.83) as

$$h_i(\tilde{x}) = \tilde{a}_i \tilde{x}_\ell^T \sum_{\substack{j=1 \\ j \neq i, \ell}}^s \tilde{A}_{ij} \tilde{x}_j. \quad (3.84)$$

The interactions  $h_i(\tilde{x})$  can be bounded as

$$\|h_i(\tilde{x})\| \leq v_{0\ell} \sum_{\substack{j=1 \\ j \neq i, \ell}}^s \epsilon_{ij} \|\tilde{x}_j\|, \quad \forall \tilde{x} \in r \quad (3.85)$$

on the region

$$r = \{\tilde{x} \in R^n: \|\tilde{x}_i\| < v_{0i}, i = 1, 2, \dots, s\}, \quad (3.86)$$

where  $v_{0i}$  are positive yet unspecified constants, and  $\epsilon_{ij} = (\tilde{a}_i^T \tilde{a}_i)^{\frac{1}{2}} \lambda_M^{\frac{1}{2}}(\tilde{A}_{ij}^T \tilde{A}_{ij})$ .

The aggregate  $s \times s$  matrix  $\tilde{W} = (\tilde{w}_{ij})$  which corresponds to the system (3.83) and constraints (3.85), is obtained following reference [17],

$$\tilde{W} = D\tilde{W} \quad (3.87)$$

where

$$D = \text{diag}\{v_{02}, v_{03}, \dots, v_{0s}, v_{01}\} \quad (3.88)$$

and the  $s \times s$  matrix  $\bar{W} = (\bar{w}_{ij})$  is defined by

$$\bar{w}_{ij} = -\delta_{ij} v_{0\ell}^{-1} \pi_i + (1 - \delta_{ij}) \bar{\xi}_{ij} \quad (3.89)$$

with  $\pi_i$  defined in (3.9).

From (3.87), it follows that  $\bar{W}$  satisfies inequalities (3.21) if and only if  $\bar{W}$  does. Inequalities (3.21) applied to  $\bar{W}$  determine the constants  $v_{01}, v_{02}, \dots, v_{0s}$  in (3.85). It is possible to calculate these constants recursively. To see this, we note that the  $k$ -th leading principal  $k \times k$  submatrix  $\bar{W}_k$  can be expressed as

$$\bar{W}_k = \begin{bmatrix} \bar{W}_{k-1} & f_k \\ g_k^T & \bar{w}_{kk} \end{bmatrix} = \begin{bmatrix} I & 0 \\ g_k^{T\bar{W}_{k-1}^{-1}} & 1 \end{bmatrix} \begin{bmatrix} \bar{W}_{k-1} & 0 \\ 0 & \bar{w}_{kk} - g_k^{T\bar{W}_{k-1}^{-1}} f_k \end{bmatrix} \begin{bmatrix} I & \bar{W}_{k-1}^{-1} f_k \\ 0 & 1 \end{bmatrix} \quad (3.90)$$

Therefore, the  $k$ -th leading principle minor of  $\bar{W}$  is

$$\det \bar{W}_k = \det \bar{W}_{k-1} (\bar{w}_{kk} - g_k^{T\bar{W}_{k-1}^{-1}} f_k) \quad (3.91)$$

For the inequalities (3.91) to be satisfied by  $\bar{W}$ , it is necessary and sufficient that

$$-\bar{w}_{kk} + g_k^{T\bar{W}_{k-1}^{-1}} f_k > 0, \quad k = 1, 2, \dots, s. \quad (3.92)$$

From (3.89), we have

$$f_k^T = (\bar{\xi}_{1k}, \bar{\xi}_{2k}, \dots, \bar{\xi}_{sk}), \quad g_k^T = (\bar{\xi}_{k1}, \bar{\xi}_{k2}, \dots, \bar{\xi}_{ks}) \quad (3.93)$$

and from (3.89) and (3.92), we get the constants  $v_{0\ell}$  as

$$v_{0\ell} < -\pi_k (g_k^{T\bar{W}_{k-1}^{-1}} f_k)^{-1}, \quad \ell = \begin{cases} 1, & k = s \\ k+1, & k \neq s \end{cases} \quad (3.94)$$

Once the constants  $v_{0i}$  are calculated by (3.94), the region  $\tilde{\Omega}$  of (3.86) is determined. Now, it remains to imbed a Liapunov function  $V: \mathbb{R}^n \rightarrow \mathbb{R}_+$  inside the region  $\Omega$  and determine a region of stability [17]

$$\tilde{\Omega} = \{\tilde{x} \in \mathbb{R}^n: V(\tilde{x}) < \gamma\} . \quad (3.95)$$

In (3.95), we choose

$$V(\tilde{x}) = \sum_{i=1}^s d_i |v_i| , \quad (3.96)$$

where  $d_i$  are positive numbers, and  $v_i = v_i(\tilde{x}_i) = \|\tilde{x}_i\|$ . Following [17], we calculate the positive constant  $\gamma$  in (3.95) using (3.94) and

$$\gamma = \min_i d_i v_{0i} , \quad i = 1, 2, \dots, s \quad (3.97)$$

where the positive vector  $d^T = (d_1, d_2, \dots, d_s)$  is computed by

$$d^T = -c^T W^{-1} , \quad (3.98)$$

where  $c$  is any positive  $s$  vector ( $c > 0$ ).

Since  $\tilde{x}_i = T_i^{-1} x_i$ , and  $\|\tilde{x}_i\| \leq \|T_i^{-1}\| \|x_i\|$ , from (3.96) and (3.97), we get finally the region of stability  $\Omega$  in the original state space, which is

$$\Omega = \{x \in \mathbb{R}^n: \sum_{i=1}^s d_i \|T_i^{-1}\| \|x_i\| < \gamma\} . \quad (3.99)$$

Now, we consider the nonlinear model of the LST given in Section 2, which belongs to the class of systems described by (2.79) with

$$A_i = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} , \quad A_{ij} = \begin{bmatrix} 0 & 0 \\ 0 & 1 \end{bmatrix} , \quad a_i = \begin{bmatrix} 0 \\ -\alpha_i \end{bmatrix} , \quad b_i = \begin{bmatrix} 0 \\ \beta_i \end{bmatrix} ,$$

$$i = 1, 2, 3 . \quad (3.100)$$

Applying the control law

$$u_i = -k_i^T x_i, \quad i = 1, 2, 3 \quad (3.101)$$

where

$$k_i^T = \beta_i^{-1} \bar{k}_i^T, \quad i = 1, 2, 3 \quad (3.102)$$

and  $\bar{k}_i^T = (\bar{k}_{i1}, \bar{k}_{i2})$ , we obtain the closed-loop uncoupled subsystems (3.81) with

$$A_i - b_i k_i^T = \begin{bmatrix} 0 & 1 \\ -\bar{k}_{i1} & -\bar{k}_{i2} \end{bmatrix}, \quad i = 1, 2, 3. \quad (3.103)$$

The gains  $\bar{k}_i$  are chosen so that each subsystem has a set of eigenvalues

$$L_i = \{-\sigma_1^i, -\sigma_2^i\}, \quad i = 1, 2, 3. \quad (3.104)$$

To get the transformed system corresponding to (3.83), we use the transformation matrix

$$T_i = \begin{bmatrix} 1 & 1 \\ -\sigma_1^i & -\sigma_2^i \end{bmatrix}, \quad i = 1, 2, 3 \quad (3.105)$$

and get

$$A_i = \begin{bmatrix} -\sigma_1^i & 0 \\ 0 & -\sigma_2^i \end{bmatrix}, \quad h_i(\tilde{x}_i) = \tilde{a}_i \tilde{x}_i^T \tilde{A}_{ij} \tilde{x}_j \\ i, j = 1, 2, 3, i \neq j; \quad \ell = \begin{cases} i+1, & i \neq 3 \\ 1, & i=3 \end{cases} \quad (3.106)$$

To compute  $\tilde{\xi}_{ij}$ , we choose  $\sigma_1^i = \sigma_1$ ,  $\sigma_2^i = \sigma_2$ ,  $i = 1, 2, 3$ , and calculate  $\|A_{ij}\| = (\sigma_1)^2 + (\sigma_2)^2$ ,  $(\tilde{a}_i^T \tilde{a}_i)^{\frac{1}{2}} = \sqrt{2} |\alpha_i| (|\sigma_1 - \sigma_2|)^{-1}$ . We can minimize the numbers  $\tilde{\xi}_{ij}$  with respect to the distance  $\rho = \sigma_2 - \sigma_1$  between the two subsystem eigenvalues. This yields

$$\tilde{\xi}_{ij} = \sqrt{2} |\alpha_i| \rho^{-1} [(\sigma_1)^2 + (\sigma_1 + \rho)^2], \quad (3.107)$$

and we get the minimal values  $\tilde{\xi}_{ij}^m$  for  $\tilde{\xi}_{ij}$  as

$$\tilde{\xi}_{ij}^m = (4 + 2\sqrt{2}) |\alpha_i| \sigma_1, \quad (3.108)$$

which is obtained for  $\rho = \sqrt{2} \sigma_1$ .

The corresponding matrix  $\bar{W}$  in (3.87), is

$$\bar{W} = \begin{bmatrix} \frac{\sigma_1}{v_{02}} & 0 & \tilde{\xi}_{13} \\ \tilde{\xi}_{21} & -\frac{\sigma_1}{v_{03}} & 0 \\ 0 & \tilde{\xi}_{32} & -\frac{\sigma_1}{v_{01}} \end{bmatrix}. \quad (3.109)$$

From (3.94) and (3.109), we get

$$v_{01} v_{02} v_{03} < \frac{(\sigma_1)^3}{\tilde{\xi}_{13} \tilde{\xi}_{21} \tilde{\xi}_{32}}. \quad (3.110)$$

Choosing  $v_{01} = v_{02} = v_{03} = v_0$ , and using (3.108) and (3.110), we compute  $v_0 < 0.584$ . Selecting  $v_0 = 0.574$ ,  $\sigma_1 = 10$ , and choosing  $c = (1, 1, 1)^T$ , we further compute from (3.98) the vector

$$d = (4.8, 13.7614, 4.2963)^T. \quad (3.111)$$

From (3.97), we calculate

$$\gamma = 2.4663, \quad (3.112)$$

and the region  $\tilde{\Omega}$  in the transformed state space as

$$\tilde{\Omega} = \{\tilde{x} \in \mathbb{R}^n: 4.8|\tilde{x}_1| + 13.7614|\tilde{x}_2| + 4.2963|\tilde{x}_3| < 2.4663\}. \quad (3.113)$$

In the original space, the stability region  $\Omega$  is finally obtained as

$$\Omega = \{x \in \mathbb{R}^n: 4.8||x_1|| + 13.7614||x_2|| + 4.2963||x_3|| < 1.3331\} \quad (3.114)$$

where we used  $||T_i^{-1}|| = 1.8500$ ,  $i = 1, 2, 3$ .

The feedback gains that yield the region  $\Omega$  are computed from (3.102) and

$$\bar{k}_i^T = (\sigma_1\sigma_2, \sigma_1 + \sigma_2)^T = (141.4213, 34.1421)^T, \quad i = 1, 2, 3 \quad (3.115)$$

as

$$\begin{aligned} k_1^T &= (1.6517, \quad 0.3988)^T \\ k_2^T &= (10.3303, \quad 2.4939)^T \\ k_3^T &= (10.7056, \quad 2.5846)^T . \end{aligned} \quad (3.116)$$

This completes the design of the LST control system.

#### 4. OPTIMAL CONTROL

In this section we will describe the application of a recently developed multilevel optimal control scheme [9] for the decentralized regulation of the LST. Such multilevel control schemes are quite efficient in the analysis of large-scale systems that may be decomposed into a number of interconnected subsystems of smaller dimensions. Since our model for the LST, described in Section 2, is a nonlinear interconnected system composed of three linear subsystems describing the motions of the spacecraft along the three axes, generation of the necessary control scheme basing the analysis on the subsystems is highly desirable in view of the complexities involved in the optimization of a nonlinear system of a large dimension. In the sequel, we will describe the general theory for the multilevel optimal control of interconnected systems, which will be followed by the specific application to the LST.

##### 4.1. Problem Formulation

Let us consider a continuous dynamic system described by the differential equation

$$\dot{x} = f(x, u) \quad (4.1)$$

where  $x(t) \in \mathbb{R}^n$  is the state and  $u(t) \in \mathbb{R}^m$  is the control function of the system at time  $t \in T$ . The function  $f: \mathbb{R}^n \times \mathbb{R}^m \rightarrow \mathbb{R}^n$  is continuous on a bounded region  $\mathcal{D} \subset \mathbb{R}^n$  and is locally Lipschitzian with respect to  $x$  in  $\mathcal{D}$  so that for every fixed control function  $u(t)$ , a unique solution  $x(t; t_0, x_0)$  exists for all initial conditions  $(t_0, x_0) \in \mathbb{R} \times \mathcal{D}$  and all  $t \in T$ ,  $T$  being an interval  $[t_0, \infty)$  of  $\mathbb{R}$ .

We assume that system (4.1) can be decomposed into  $s$  interconnected subsystems

$$\dot{x}_i = A_i x_i + B_i u_i + h_i(x) \quad , \quad i = 1, 2, \dots, s \quad (4.2)$$

where,  $x_i \in R^{n_i}$  is the state of the  $i$ -th subsystem so that

$$R^n = R^{n_1} \times R^{n_2} \times \dots \times R^{n_s};$$

$u_i \in R^{m_i}$  is the control function of the  $i$ -th subsystem so that

$$R^m = R^{m_1} \times R^{m_2} \times \dots \times R^{m_s};$$

$A_i \in R^{n_i \times n_i}$  and  $B_i \in R^{n_i \times m_i}$  are constant matrices; and  $h_i: R^n \rightarrow R^{n_i}$  is the function which represents the interconnection of the  $i$ -th subsystem inside the overall system.

The multilevel control scheme [9] used for the optimization of system (4.2) can be developed by considering the control function  $u_i(t)$  as consisting of two parts, the local control  $u_i^l(t)$  and the global control  $u_i^g(t)$ ,

$$u_i(t) = u_i^l(t) + u_i^g(t). \quad (4.3)$$

The local control  $u_i^l(t)$  is chosen as a linear control law

$$u_i^l(t) = -K_i^l x_i(t) \quad (4.4)$$

to optimize isolated subsystems, and the global control law  $u_i^g(t)$  is chosen as a suitable function of the state

$$u_i^g(t) = -K_i^g(x(t)) \quad (4.5)$$

to minimize the performance deviation from the optimum due to the presence of interconnections among the subsystems.

With the application of the control (4.3), the equations (4.2) governing the system under consideration take the form,

$$\dot{x}_i = A_i x_i + B_i u_i^l + h_i(x) + B_i u_i^g, \quad i = 1, 2, \dots, s. \quad (4.6)$$



Since, as described earlier, the global control functions  $u_i^g(t)$  are assigned only the task of reducing the effects of interconnections  $h_i(x)$  the terms

$$h_{e_i}(x, u_i^g) = h_i(x) + B_i u_i^g, \quad i = 1, 2, \dots, s, \quad (4.7)$$

may be regarded as the "effective interconnections" among the  $s$  isolated subsystems

$$\dot{x}_i = A_i x_i + B_i u_i^k, \quad i = 1, 2, \dots, s. \quad (4.8)$$

We shall assume that all  $s$ -pairs  $(A_i, B_i)$  are completely controllable, and that with each isolated subsystem (4.8) a quadratic performance index

$$J_i(t_0, x_{i0}, u_i^k) = \int_{t_0}^{\infty} \{ \|x_i(t)\|_{Q_i}^2 + \|u_i^k(t)\|_{R_i}^2 \} dt \quad (4.9)$$

is associated. In (4.9)  $Q_i \in \mathbb{R}^{n_i \times n_i}$  is a symmetric nonnegative definite matrix and  $R_i \in \mathbb{R}^{m_i \times m_i}$  is a symmetric positive definite matrix.

The local control  $u_i^k(t)$  in (4.4) can now be chosen to minimize the performance index  $J_i(t_0, x_{i0}, u_i)$  in (4.9). From linear-quadratic regulator theory [18], the optimal control  $u_i^{k*}(t)$  is given by

$$u_i^{k*}(t) = -K_i^{k*} x_i(t) \quad (4.10)$$

where

$$K_i^{k*} = R_i^{-1} B_i^T P_i. \quad (4.11)$$

In (4.11),  $P_i \in \mathbb{R}^{n_i \times n_i}$  is symmetric and is the positive definite solution of the algebraic Riccati equation

$$P_i A_i + A_i^T P_i - P_i B_i R_i^{-1} B_i^T P_i + Q_i = 0. \quad (4.12)$$

The optimal cost  $J_i^*(t_0, x_{i0}) = J_i(t_0, x_{i0}, u_i^{k*})$  can in this case be calculated

as

$$J_i^*(t_0, x_{i0}) = \|x_{i0}\|_{P_i}^2 \quad (4.13)$$

Furthermore, under the assumption that  $Q_i$  can be factored as  $Q_i = C_i C_i^T$ , where  $C_i \in \mathbb{R}^{n_i \times n_i}$  such that the pair  $(A_i, C_i)$  is completely observable, each closed-loop subsystem

$$\dot{x}_i = (A_i - B_i R_i^{-1} B_i^T P_i) x_i, \quad i = 1, 2, \dots, s, \quad (4.14)$$

is globally asymptotically stable.

The controls  $u_i^{g*}(t)$ ,  $i = 1, 2, \dots, s$ , will not, in general, be optimal for the composite system (4.6) and will not result in the optimal cost

$$J^*(t_0, x_0) = \sum_{i=1}^s J_i^*(t_0, x_0) \quad (4.15)$$

unless the effective interconnection functions  $h_{e_i}(x, u_i^g) \equiv 0$ . When  $h_{e_i}(x, u_i^g) \neq 0$ , the controls  $u_i^{g*}(t)$  produce a value of the performance index for the composite system given by

$$\tilde{J}(t_0, x_0) = \sum_{i=1}^s \tilde{J}_i(t_0, x_{i0}) \quad (4.16)$$

where

$$\tilde{J}_i(t_0, x_{i0}) = \tilde{J}_i(t_0, x_{i0}, u_i^{g*}) \quad (4.16)$$

It is obvious that

$$\tilde{J}(t_0, x_0) \geq J^*(t_0, x_0) \quad \forall (t_0, x_0) \in T \times \mathbb{R}^n \quad (4.17)$$

and the local control law  $u_i^{g*}(t)$  in (4.10) can only be a suboptimal policy for the composite system (4.6), with an index of suboptimality  $\epsilon > 0$  defined by the inequality

$$\tilde{J}(t_0, x_0) \leq (1+\epsilon) J^*(t_0, x_0) \quad \forall (t_0, x_0) \in T \times \mathbb{R}^n \quad (4.18)$$

The suboptimality index  $\epsilon$  for the system with the optimal local control,

$$\dot{x}_i = (A_i - B_i R_i^{-1} B_i^T P_i) x_i + h_{e_i}(x, u_i^g), \quad i = 1, 2, \dots, s \quad (4.19)$$

depends on the size of the effective interactions  $h_{e_i}(x, u_i^g)$  and hence is a measure of the performance deterioration due to these.

We can now give a formal definition of this concept.

*Definition.* The system (4.19) with the optimal local control law (4.10) is said to be suboptimal with the index  $\epsilon$  if there exists a number  $\epsilon > 0$  for which inequality (4.18) is satisfied.

As described earlier, the suboptimality index  $\epsilon$  is a function of the interactions  $h_{e_i}(x, u_i^g)$  and the following problem is of interest:

*Problem 1.* Establish conditions on  $h_{e_i}(x, u_i^g)$  to guarantee a prescribed value of the suboptimality index  $\epsilon$ .

It is important to note that in Problem 1, the rate of the global control function  $u_i^g(t)$  is ignored as it is taken together with the existing interconnections  $h_i(x)$  in the system to yield the effective interconnection function  $h_{e_i}(x, u_i^g)$ . However, as we shall see later, the solution to Problem 1 indicates a method of choosing the global control  $u_i^g(t)$  so as to reduce the size of  $h_{e_i}(x, u_i^g)$  and, hence, minimize the suboptimality index  $\epsilon$ . In other words, we consider the index  $\epsilon = \epsilon[||h_e(x, u^g)||]$  where  $h_e: \mathbb{R}^n \times \mathbb{R}^m \rightarrow \mathbb{R}^n$  is  $h_e = [h_{e_1}^T, h_{e_2}^T, \dots, h_{e_s}^T]^T$  and  $u^g \in \mathbb{R}^m$  is  $u^g = [(u_1^g)^T, (u_2^g)^T, \dots, (u_s^g)^T]^T$  and solve the following:

*Problem 2.* Find a control law of the form (4.5) or equivalently,

$$u^g(t) = -K^g(x(t)) \quad (4.20)$$

for which

$$\epsilon^* = \inf_{u^G(t)} \epsilon \{ ||| h_e(x, u^G) ||| \} \forall x \in \mathcal{D} \quad (4.21)$$

is attained.

We will now provide the solutions to the above problems.

#### 4.2. Multilevel Optimization

A solution to Problem 1 may be obtained by using the classical Hamilton-Jacobi theory. Since in our optimization procedure, we chose the local control laws (4.10) to optimize the decoupled subsystems, the optimal indices satisfy the corresponding Hamilton-Jacobi equations. When the subsystems are interconnected, the equations are not satisfied by the respective performance indices and the overall system is not optimal. However, a majorization procedure is possible to provide an estimate of the performance deviation from the optimum due to the interactions.

Now, we provide a solution to Problem 1 by the following:

Theorem 4.1. Let there exist nonnegative numbers  $\xi_{ij}$  such that the function  $h_{e_i}(x, u_i^G)$  in (4.19) satisfy the constraints

$$||| h_{e_i}(x, u_i^G) ||| \leq \sum_{j=1}^s \xi_{ij} ||| x_j |||, \forall x \in \mathbb{R}^n, \forall i = 1, 2, \dots, s \quad (4.22)$$

and

$$\xi \leq \frac{\epsilon}{1+\epsilon} \frac{\lambda_m(W)}{2\lambda_M(P)} \quad (4.23)$$

where  $\xi = \sum_{i=1}^s \sum_{j=1}^s \xi_{ij}$ ,  $P = \text{diag}\{P_1, P_2, \dots, P_s\}$ ,  $W = \text{diag}\{W_1, W_2, \dots, W_s\}$ ,  $P_i$  being defined by (4.12) and  $W_i = Q_i + P_i B_i R_i^{-1} B_i^T P_i$ , and  $\lambda_M(P)$  and  $\lambda_m(W)$  are the maximum and minimum eigenvalues of  $P$  and  $W$  respectively. Then the composite system (4.19) is

(i) suboptimal with index  $\epsilon$

and

(ii) globally asymptotically stable.

*Proof.* Since the decoupled subsystems (4.14) are optimal, the functions  $v_i(x_i) = ||x_i||_{P_i}^2$ ,  $i = 1, 2, \dots, s$ , satisfy individually the Hamilton-Jacobi equations

$$\begin{aligned} & [\text{grad } v_i(x_i)]^T [(A_i - B_i K_i^{l*}) x_i] \\ & + ||x_i||_{Q_i}^2 + ||K_i^{l*} x_i||_{R_i}^2 = 0, \\ & \forall x_i \in R^{n_i}, i = 1, 2, \dots, s. \end{aligned} \quad (4.24)$$

Now, the time-derivative  $\dot{v}_i(x_i)$  can be calculated along the trajectories  $\tilde{x}_i(t)$  of the composite system (4.19) as

$$\dot{v}_i(\tilde{x}_i) = [\text{grad } v_i(\tilde{x}_i)]^T \{ (A_i - B_i R_i^{-1} B_i^T P_i) \tilde{x}_i + h_{e_i}(\tilde{x}, u_i^g) \} \quad (4.25)$$

where  $\tilde{x} = [\tilde{x}_1^T, \tilde{x}_2^T, \dots, \tilde{x}_s^T]^T$ .

Substitution of (4.25) in (4.24) and rearrangement of terms gives

$$\begin{aligned} ||\tilde{x}_i||_{W_i}^2 &= - (1+\epsilon) \dot{v}_i(x_i) + (1+\epsilon) [\text{grad } v_i(\tilde{x}_i)]^T h_{e_i}(\tilde{x}, u_i^g) \\ &- \epsilon ||\tilde{x}_i||_{W_i}^2, \quad \forall \tilde{x} \in R^n, i = 1, 2, \dots, s, \end{aligned} \quad (4.26)$$

where the simplification  $||\tilde{x}_i||_{Q_i}^2 + ||K_i^{l*} x_i||_{R_i}^2 = ||\tilde{x}_i||_{W_i}^2$  with  $W_i = Q_i + P_i B_i R_i^{-1} B_i^T P_i$  is made.

Denoting  $v(\tilde{x}) = \sum_{i=1}^s v_i(\tilde{x}_i)$  and summing the s-equations in (4.26) we get,

$$\begin{aligned} ||\tilde{x}||_W^2 &= - (1+\epsilon) \dot{v}(\tilde{x}) + (1+\epsilon) [\text{grad } v(\tilde{x})]^T h_e(\tilde{x}, u^g) \\ &- \epsilon ||\tilde{x}||_W^2, \quad \forall \tilde{x} \in R^n. \end{aligned} \quad (4.27)$$

Now, integrating (4.27) from  $t_0$  to  $\infty$  we obtain

$$\begin{aligned} \tilde{J}(t_0, x_0) &= (1+\varepsilon) J^*(t_0, x_0) \\ &+ (1+\varepsilon) \int_{t_0}^{\infty} \{ [\text{grad } v(\tilde{x})]^T h_e(\tilde{x}, u^g) - \frac{\varepsilon}{1+\varepsilon} \|\tilde{x}\|_W^2 \} dt, \end{aligned} \quad (4.28)$$

where  $\tilde{J}$  and  $J^*$  are defined in (4.15) and (4.16).

It is now simple to observe from (4.18) and (4.28) that the system is suboptimal with index  $\varepsilon$  if

$$\int_{t_0}^{\infty} \{ [\text{grad } v(\tilde{x})]^T h_e(\tilde{x}, u^g) - \frac{\varepsilon}{1+\varepsilon} \|\tilde{x}\|_W^2 \} dt \geq 0, \quad \forall \tilde{x} \in \mathbb{R}^n. \quad (4.29)$$

For further simplification of (4.29) we note that

$$v(\tilde{x}) = \sum_{i=1}^S v_i(\tilde{x}_i) = \sum_{i=1}^S \|\tilde{x}_i\|_{p_i}^2 = \|\tilde{x}\|_p^2. \quad (4.30)$$

Also, since  $\|h_{e_i}(\tilde{x}, u_i^g)\| \leq \sum_{j=1}^S \xi_{ij} \|\tilde{x}_j\|, \forall \tilde{x} \in \mathbb{R}^n$  we have the inequality

$$\|h_e(\tilde{x}, u^g)\| \leq \xi \|\tilde{x}\|, \quad \forall \tilde{x} \in \mathbb{R}^n \quad (4.31)$$

where  $\xi = \sum_{i=1}^S \sum_{j=1}^S \xi_{ij}$ .

Using (4.30) and (4.31) it can be easily shown that a sufficient condition for the inequality (4.29) to hold is

$$2\varepsilon p \xi \|\tilde{x}\| \leq \frac{\varepsilon}{1+\varepsilon} \|\tilde{x}\|_W^2, \quad \forall \tilde{x} \in \mathbb{R}^n \quad (4.32)$$

which, however, is implied by the main inequality (4.23) of the Theorem.

To complete the proof of the Theorem, we demonstrate the global asymptotic stability of the system (4.19) by using the function  $v(\tilde{x}) = \|\tilde{x}\|_p^2$  as a

Liapunov function. Note that  $v(\tilde{x})$  is positive definite since  $P$  is a diagonal matrix formed from the positive definite solutions of the  $s$  Riccati equations (4.12). Further, the time-derivative of  $v(\tilde{x})$  along the solutions of (4.19) gives

$$\dot{v}(\tilde{x}) = - \|\tilde{x}\|_W^2 + 2\tilde{x}^T P h_e(\tilde{x}, u^G) \leq 0 \quad \forall \tilde{x} \in \mathbb{R}^n, \quad (4.33)$$

from (4.31) and (4.23). This completes the proof of the Theorem.

It is important to note that the above theorem provides an explicit algebraic constraint on the interactions that is easy to check. Inequality (4.23) involves calculation of eigenvalues of block-diagonal matrices  $P$  and  $W$ , and since  $\lambda_M(P) = \max_i \{\lambda_M(P_i)\}$  and  $\lambda_m(W) = \min_i \{\lambda_m(W_i)\}$ , the calculation can be carried out on the subsystem level.

In the context of the above Theorem, it is of interest to consider Problem 2 of determining the global control  $u^G(t)$  so as to minimize the suboptimality index  $\epsilon$ . From (4.23) and (4.31), it is evident that  $\epsilon$  is a non-decreasing function of  $\|h_e(x, u^G)\|$  and hence, Problem 2 reduces to one of choosing  $u^G(t)$  to minimize  $\|h_e(x, u^G)\|$ . This function minimization problem is particularly simple in the present case since, from (4.7)

$$h_e(x, u^G) = h(x) + Bu^G \quad (4.34)$$

which, on using the control law (4.5) reduces to

$$h_e(x, u^G) = h(x) - BK^G(x) \quad (4.35)$$

A perfect neutralization of the effects of interconnections occurs if a choice of  $K^G(x)$  results in

$$BK^G(x) = -h(x) \quad (4.36)$$

and, in this case,  $\epsilon = 0$ . In the special case, when  $B$  is square and non-

singular, the explicit expression for  $K^g(x)$  is available as

$$K^g(x) = -B^{-1}h(x) . \quad (4.37)$$

In general, a perfect neutralization of the interaction effects mentioned above, is not possible and one may attempt to minimize  $||h(x) - BK^g(x)||$  by the proper choice of  $K^g(x)$  in order to solve Problem 2. This is admittedly a complex minimization problem and a general solution is difficult to obtain. However, in the particular case of linear interconnections, the problem can be simplified and an elegant solution can be provided. This is, we assume

$$\hat{h}(t, x) = Hx \quad (4.38)$$

where  $H \in \mathbb{R}^{n \times n}$ . In this case, the global control can also be chosen as a linear law

$$K^g(x) = K^g x \quad (4.39)$$

where  $K^g \in \mathbb{R}^{m \times n}$ . With (52) and (53), Problem 2 simplifies to:

Problem 2'. Choose the matrix  $K^g$  such that  $\inf_{K^g} ||(H-BK^g)x||$  is achieved for all  $x \in \mathbb{R}^n$ .

Remembering that  $||(H-BK^g)x|| \leq ||H-BK^g|| ||x||$  holds for all  $x \in \mathbb{R}^n$ , Problem 2' actually reduces to finding  $\min_{K^g} ||H-BK^g||$ . When  $\text{rank } B = m$ , the solution to this latter problem is well-known and  $K^g$  is given by

$$K^g = (B^T B)^{-1} B^T H \quad (4.40)$$

where  $(B^T B)^{-1} B^T$  is the Moore-Penrose generalized inverse of  $B$  [15]. It is interesting to note that in the particular case when

$$\text{Rank } [B \mid H] = \text{Rank } B \quad (4.41)$$



the choice (4.40) leads to a perfect neutralization of interaction effects and  $\varepsilon = 0$ .

### 4.3. An Illustrative Example

For the purpose of illustrating the multilevel control scheme presented here, let us consider the following example.

The system is described by

$$\dot{x} = Ax + Bu \quad (4.42)$$

where

$$A = \begin{bmatrix} -5 & 6 & 0 & -0.0095 \\ 4 & -4 & 0.003 & 0 \\ -0.00332 & 0 & -3 & 1 \\ 0 & 0.00995 & 8 & -2 \end{bmatrix} \quad \text{and} \quad B = \begin{bmatrix} 1 & 0 \\ 0 & 0 \\ 0 & 0 \\ 0 & 1 \end{bmatrix}$$

and is required to be optimized with respect to the performance index

$$J = \int_{t_0}^{\infty} \{ ||x||^2 + ||u||^2 \} dt . \quad (4.43)$$

In this particular case, the dimension of the system ( $n = 4$ ) is small and hence the problem is amenable for a direct analysis and the required control can be obtained from solving the associated Riccati equation (of fourth order). However, as our interest here is to provide an illustration of the decentralized optimal control scheme<sup>\*</sup>, let us consider system (4.42) as being

---

\* Besides the advantage of permitting an analysis based on the subsystems of small orders, the decentralized control scheme presented results in important connectivity properties of the system. The suboptimality and stability of the system remain invariant under structural perturbations caused by the on-off participation of the parts of the system. This property, however, does not result when direct optimization of the system is carried out [10].

composed of two subsystems

$$\dot{x}_1 = \begin{bmatrix} -5 & 6 \\ 4 & -4 \end{bmatrix} x_1 + \begin{bmatrix} 1 \\ 0 \end{bmatrix} u_1 \quad (4.44)$$

and

$$\dot{x}_2 = \begin{bmatrix} -3 & 1 \\ 8 & -2 \end{bmatrix} x_2 + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u_2 \quad (4.45)$$

with the interconnection matrix

$$H = \begin{bmatrix} 0 & 0 & 0 & -0.0095 \\ 0 & 0 & 0.003 & 0 \\ -0.00332 & 0 & 0 & 0 \\ 0 & 0.00995 & 0 & 0 \end{bmatrix} \quad (4.46)$$

By splitting the control functions  $u_1$  and  $u_2$  into a local component and a global component, the decoupled subsystems (4.44) and (4.45) can be optimized with respect to the performance indices

$$J_1 = \int_{t_0}^{\infty} \{ ||x_1||^2 + ||u_1||^2 \} dt \quad \text{and} \quad J_2 = \int_{t_0}^{\infty} \{ ||x_2||^2 + ||u_2||^2 \} dt. \quad (4.47)$$

The solutions of the associated Riccati equations can be obtained as

$$P_1 = \begin{bmatrix} 1.1910 & 1.5411 \\ 1.5411 & 2.1397 \end{bmatrix} \quad \text{and} \quad P_2 = \begin{bmatrix} 5.5591 & 2.3746 \\ 2.3746 & 1.1224 \end{bmatrix}$$

and the local control laws are,

$$\begin{aligned} u_1 &= -[1.1910 \quad 1.5411] x_1 \\ u_2 &= -[2.3746 \quad 1.1224] x_2 \end{aligned} \quad (4.48)$$

In the absence of the global controls, the functions (4.48) will only be suboptimal policies for the overall system (4.42) with the index of sub-

optimality  $\epsilon$  given by

$$\|H\| \leq \frac{\epsilon}{\epsilon+1} \cdot \frac{\min\{\lambda_m(W_1), \lambda_m(W_2)\}}{2 \max\{\lambda_M(P_1), \lambda_M(P_2)\}} \quad (4.49)$$

where  $W_i = Q_i + P_i B_i R_i^{-1} B_i^T P_i$ ,  $i = 1, 2$ , are

$$W_1 = \begin{bmatrix} 2.4185 & 1.8354 \\ 1.8354 & 3.3748 \end{bmatrix} \quad \text{and} \quad W_2 = \begin{bmatrix} 6.6386 & 2.6651 \\ 2.6651 & 2.2597 \end{bmatrix} .$$

Inequality (4.49) is satisfied with  $\epsilon = 2$  and hence the performance degradation from the optimum is 200% .

In order to improve the performance, we now use the global controls  $u_1^g$  and  $u_2^g$  given by

$$u^g = - (B^T B)^{-1} B^T H x \quad (4.50)$$

where  $u^g = \begin{bmatrix} u_1^g \\ u_2^g \end{bmatrix}$ . (4.50) can be simplified to yield

$$\begin{aligned} u_1^g &= - [ 0 \quad -0.0095 ] x_2 \\ u_2^g &= - [ 0 \quad 0.00995 ] x_1 . \end{aligned} \quad (4.51)$$

The effective interaction matrix  $\tilde{H}$  with the application of the global control is

$$\begin{aligned} \tilde{H} &= [ I - B(B^T B)^{-1} B^T ] H \\ &= \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & 0 & 0.003 & 0 \\ -0.00332 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} \end{aligned} \quad (4.52)$$

and the suboptimality inequality (4.49) can now be satisfied with a value  $\epsilon = 0.2$ . Hence the performance degradation is reduced from the original 200% to only 20%, thus illustrating the effectiveness of the global controls.

#### 4.4. Application to LST

The results developed in the earlier parts of this section may be directly used for the multilevel optimization of the LST. As described in Section 2, the model for the LST is a set of three interconnected subsystems, described by (2.20) as,

$$\dot{x}_i = A_i x_i + b_i u_i + h_i(x), \quad i = 1, 2, 3, \quad (4.53)$$

where  $x_i \in R^2$ ,  $A_i = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix}$ ,  $b_i = \begin{bmatrix} 0 \\ \beta_i \end{bmatrix}$  and  $h_1(x) = \begin{bmatrix} 0 \\ -\alpha_2 x_{32} x_{12} \end{bmatrix}$ ,

$h_2(x) = \begin{bmatrix} 0 \\ -\alpha_2 x_{32} x_{12} \end{bmatrix}$ ,  $h_3(x) = \begin{bmatrix} 0 \\ -\alpha_3 x_{12} x_{22} \end{bmatrix}$ ,  $x$  being the composite state-

vector  $x = [x_1^T, x_2^T, x_3^T]^T$  and  $x_i = [x_{i1}, x_{i2}]^T$ ,  $i = 1, 2, 3$ .

Following our multilevel control policy, we split each of the control functions  $u_i$  into a local component  $u_i^l$  and a global component  $u_i^g$  and optimize the decoupled subsystems

$$\dot{x}_i = A_i x_i + b_i u_i^l, \quad i = 1, 2, 3, \quad (4.54)$$

with respect to the performance indices

$$J_i = \int_{t_0}^{\infty} \{ ||x_i||^2 + ||u_i^l||^2 \} dt, \quad i = 1, 2, 3, \quad (4.55)$$

obtained with the choice  $Q_i = I_{2 \times 2}$  and  $R_i = 1 \quad \forall i = 1, 2, 3$ . The solution of this linear-quadratic optimal control problem is simple and involves

the solution of the associated Riccati equations,

$$A_i^T P_i + P_i A_i - P_i b_i b_i^T P_i + Q_i = 0, \quad i = 1, 2, 3. \quad (4.56)$$

With the specified structure of  $A_i$  and  $b_i$ , the solution of (4.56) can be obtained as,

$$P_i = \begin{bmatrix} (1 + \frac{2}{\beta_i})^{1/2} & \frac{1}{\beta_i} \\ \frac{1}{\beta_i} & \frac{1}{\beta_i} (1 + \frac{2}{\beta_i})^{1/2} \end{bmatrix} \quad (4.57)$$

and the local optimal controls are

$$u_i^l = -b_i^T P_i x_i = - [1 \quad (1 + \frac{2}{\beta_i})^{1/2}] x_i, \quad i = 1, 2, 3. \quad (4.58)$$

However, in the absence of a suitable choice of the global control functions  $u_i^g$ , (4.58) will only be suboptimal for the composite system (4.53), with the index of suboptimality  $\epsilon$  determined by the size of the effective interconnections,

$$h_{e_i}(x, u_i^g) = h_i(x) + b_i u_i^g, \quad i = 1, 2, 3. \quad (4.59)$$

(4.59) can be simplified to yield

$$\begin{aligned} h_{e_1}(x, u_1^g) &= \begin{bmatrix} 0 \\ -\alpha_1 x_{22} x_{32} + \beta_1 u_1^g \end{bmatrix} \\ h_{e_2}(x, u_2^g) &= \begin{bmatrix} 0 \\ -\alpha_2 x_{12} x_{32} + \beta_2 u_2^g \end{bmatrix} \\ h_{e_3}(x, u_3^g) &= \begin{bmatrix} 0 \\ -\alpha_3 x_{12} x_{22} + \beta_3 u_3^g \end{bmatrix} \end{aligned} \quad (4.60)$$

It is now simple to observe that the choice of

$$\begin{aligned} u_1^g(x) &= \frac{\alpha_1}{\beta_1} x_{22} x_{32} \\ u_2^g(x) &= \frac{\alpha_2}{\beta_2} x_{12} x_{32} \\ u_3^g(x) &= \frac{\alpha_3}{\beta_3} x_{12} x_{22} \end{aligned} \quad (4.61)$$

will make the functions  $h_{e_i}(x, u_1^g) \equiv 0$  and hence  $\varepsilon = 0$ , thus resulting in no degradation of the performance from the optimum.

It is of interest to evaluate the control functions for a representation set of values of the parameters of the LST. For the values of the inertia components\*  $I_x = 14656 \text{ Kg}_m^2$ ,  $I_y = 91772 \text{ Kg}_m^2$  and  $I_z = 95027 \text{ Kg}_m^2$  and typical reaction wheel constants  $K_1 = K_2 = K_3 = 12.57 \times 10^5 \text{ N-m/rad.}$ , the values of  $\alpha_i$ ,  $\beta_i$ ,  $i = 1, 2, 3$  can be calculated as

$$\begin{aligned} \alpha_1 &= 0.2221 & \beta_1 &= 85.62 \\ \alpha_2 &= -0.08754 & \text{and} & \beta_2 = 13.69 \\ \alpha_3 &= 0.8112 & \beta_3 &= 13.21 \end{aligned}$$

Hence, the control components can be evaluated from (4.58) and (4.61) as,

$$\begin{aligned} u_1^l &= -[1 \quad 1.012] \begin{bmatrix} x_{11} \\ x_{12} \end{bmatrix} \\ u_2^l &= -[1 \quad 1.061] \begin{bmatrix} x_{21} \\ x_{22} \end{bmatrix} \\ u_3^l &= -[1 \quad 1.07] \begin{bmatrix} x_{31} \\ x_{32} \end{bmatrix} \end{aligned} \quad (4.62)$$

\* These values correspond to the on-orbit configuration of the LST with extended light shield and solar wings, with the corresponding mass of the body totalling 9380 Kg [11].

and

$$\begin{aligned}u_1^g &= 0.0026 x_{22}x_{32} \\u_2^g &= -0.064 x_{12}x_{32} \\u_3^g &= 0.0613 x_{12}x_{22} \quad \cdot\end{aligned}\tag{4.63}$$

This completes the multilevel optimization of the LST control system.

## 5. CONCLUSIONS

A multilevel scheme was proposed for control of Large Space Telescope modeled by a three-axis-six-order nonlinear equation. Local controllers were used on the subsystem level to stabilize motions corresponding to the three axes. Global controllers were applied to reduce (and sometimes nullify) the interactions among the subsystems. A multilevel optimization method was developed whereby local quadratic optimizations were performed on the subsystem level, and global control was again used to reduce (nullify) the effect of interactions.

The proposed multilevel stabilization and optimization methods are presented as general tools for design and then used in the design of the LST Control System. Furthermore, the methods are entirely computerized (Appendices A.1 and 2), so that they can accommodate higher order LST models with both conceptual and numerical advantages over the standard straightforward design techniques.



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APPENDIX

COMPUTER APPLICATION

A.1. Stabilization Program

A.2. Optimization Program

## A.1. Stabilization Program

The entire stabilization method is computerized. In this section we present the computer program for the stabilization of a class of large-scale systems by local state feedback, according to Section 3.3. The program is written in FORTRAN IV, for HP2100 computer. It is, basically, an interactive user oriented program.

Designers can enter the program from VDU (Visual Display Unit) and thus freely alter the course of computation, according to the nature of the problem. Program accepts input data from a logical unit that has to be previously assigned. As a result of computation, it prints out stabilizing parameter  $\alpha$ , corresponding aggregation matrix, stabilizing set of subsystem eigenvalues, and enables the designer to reenter the program with so computed new set. The program finally prints out the corresponding subsystem feedback gains. The name of the main program is PPI. Its function is to coordinate the sequence of actions during the course of the stabilization and to enable the designer to access the program at various points during its operation. The program PPI calls subprograms, DECP, PPL, TRF, AGR and MINV. The processing of variables between the main program and subroutines is realized via COMMON block.

### Program PPI

#### *Purpose:*

Local stabilization of a class of large-scale linear systems.

#### *Description of input parameters:*

A - N by N system matrix.

B - N by M input matrix.

II - one dimensional integer array. It stores dimensions of each subsystem. The other parameters are working variables.

User has to specify integers  $N$  and  $M$  and number of subsystems  $IS$ , into which system matrix  $A$ , and input matrix  $B$  are decomposed. During the course of stabilization, user has to enter the program with subsystem eigenvalues, and specify an increment  $\delta$  by which  $\alpha$  is increased during the process of iterations.

At the very beginning of the program, the user has to assign input-output units. Also, during the operation of the program, user communicates with the program by specifying commands, by which the sequence of calculations is controlled. These commands are in the "question-answer" form. For example, program prints out the question:

"DO YOU WANT TO CONTINUE, YES OR NO". The user then types either "YES" or "NO" accordingly. Other commands are self explanatory, and are not going to be discussed here.

### Subroutine DECP

#### *Purpose:*

Decomposes system matrix  $A$  and input matrix  $B$  into subsystems. The product of the decomposition is stored in  $A2$  and  $B2$ .

#### *Usage:*

```
CALL DECP (IS, M, N)
```

#### *Description of parameters:*

- IS - Number of subsystems.
- M - Number of inputs.
- N - Order of the overall system.

The following parameters are passed via COMMON block as;

```
COMMON A, B, A2, B2, II
```

A - N by N system matrix.

- B - N by M input matrix.
- A2 - Three dimensional array which contains the product of the decomposition of the matrix A .
- B2 - Three dimensional array which contains the product of the decomposition of the matrix B .
- II - One dimensional integer array which contains the dimensions of each subsystem.

*Subroutines required:* None.

#### Subroutine PPL

*Purpose:*

Pole shifting using state feedback.

*Usage:*

CALL PPL (N, IW)

*Description of parameters:*

- N - Order of the system.
- IW - Integer for the output logical unit.

The following parameters are passed via COMMON block as:

COMMON A1, B1, B2, II A, Q1, Q, I11, B, R1, R2, D, SK

- A - N by N system matrix
- B - N-th dimensional input vector
- R1 - One dimensional array which contains real parts of eigenvalues of matrix A .
- R2 - One dimensional array which contains imaginary parts of eigenvalues of matrix A .
- D - N-th dimensional gain vector.

All other parameters are working variables, which are placed in COMMON

block in order to make it consistent with the COMMON block of the main program PPI.

*Subroutines required:* ALAM, DISP, KBAR.

As a result of the pole shifting, the subroutine passes back matrix  $A$  of the closed loop system, the gain vector  $D$ , and the new eigenvalues. The subroutine itself is written as a user-oriented interactive program. The user enters the desired eigenvalues from VDU. The commands for controlling a sequence of computations, are self explanatory.

### Subroutine TRF

*Purpose:*

Transforms subsystems by similarity transformation.

*Usage:*

CALL TRF (IS)

*Description of parameters:*

IS - Number of subsystems.

The following parameters are passed via COMMON block as:

COMMON A, B, A2, B2, II, A1, Q1, Q, I11, B3

A2 - Three dimensional array. It contains the product of the decomposition of the system matrix  $A$ .

B2 - Three dimensional array. It contains the product of the decomposition of the input matrix  $B$ .

II - One dimensional integer array that contains dimensions of each subsystem.

Q - Three dimensional array that contains transformation matrices.

All other parameters are working variables.

*Subroutines required:* MINV

The product of transformation is in A2 and B2. The array Q is unchanged.

Subroutine AGR

*Purpose:*

Forms an aggregate matrix.

*Usage:*

CALL AGR (IS)

*Description of parameters:*

IS - Number of subsystems.

The following parameters are passed via COMMON block as:

COMMON A, B, A2, B2, II, A4, A3, Q, I11, B3, R1, R2, KB, SK

A2 - Three dimensional array which contains the product of the decomposition of the matrix A .

B2 - Three dimensional array which contains the product of the decomposition of the matrix B .

II - Integer array that contains the dimensions of each subsystem.

R1 - One dimensional array that contains real parts of subsystem eigenvalues.

A3 - Matrix that contains the aggregate model.

All other parameters are working variables.

*Subroutine required:* ALAM, BIG1, SMALL.

Subroutine KBAR

*Purpose:*

Computes gain vector for state feedback control.



*Usage:*

CALL KBAR (A, N, Z, IZ, D, B)

*Description of parameters:*

- A - N by N system matrix.
- N - Dimension of the system.
- Z - One dimensional array that contains the desired characteristic polynomial.
- IZ - Its dimension.
- D - One dimensional array that contains resultant gain vector.
- B - Input vector.

*Subroutine required:* COEF1, SCALU, VECPR, MINV.

*Method:*

Described in reference [13].

Subroutine ALAM*Purpose:*

Calculates eigenvalues of general N by N matrix.

*Usage:*

CALL ALAM (A, N, D, COF, R1, R2)

*Description of parameters:*

- A - N by N system matrix.
- N - Dimension of the system.
- D - N+1 dimensional working vector.
- COF - N+1 dimensional working vector.
- R1 - One dimensional array of real parts of eigenvalues of matrix A .
- R2 - One dimensional array of imaginary parts of eigenvalues of matrix A .

*Subroutines required:* COEF1, POLRT

*Method:*

Computes coefficients of characteristic polynomial, and calculates its zeros.

Subroutine COEF1 (A, N, D)*Purpose:*

Calculates coefficients of the characteristic polynomial of matrix A .

*Usage:*

CALL COEF1 (A, N, D)

*Description of parameters:*

A - N by N system matrix.

N - Dimensions of the system.

D - One dimensional array of coefficients of characteristic polynomial.

*Subroutines required:* UNIT1, PROD1, TRAC1, SCML, ADD1

*Method:*

Uses Souriau-Frame-Faddeev algorithm.

Subroutine DISP*Purpose:*

Form polynomial from its zeros.

*Usage:*

CALL DISP (R1, R2, N, Z)

*Descriptions of parameters:*

R1 - One dimensional array of real parts of roots of a given polynomial.

R2 - One dimensional array of imaginary parts of roots of a given polynomial.

N - Order of a polynomial.

Z - One dimensional array that contains computed coefficients  
of the polynomial.

*Subroutine required:* PMPY

All other subroutines, listed in the Appendix are self explanatory and are not going to be explained here. Subroutines PMPY, POLRT and MINV are IBM-SSP subroutines.



```

DO 372 K=1,N
372 WRITE(IW,105) (B(K,J),J=1,M)
105 FORMAT(1X,5F14,6)
WRITE(IW,147)
147 FORMAT(1X,"SPECIFY NUMBER OF SUBSYSTEMS"/1X,"I2")
READ(IRD,111) IS
WRITE(IW,544) IS
544 FORMAT(1X,"NUMBER OF SUBSYSTEMS = ",I2)
WRITE(IW,149)
149 FORMAT(1X,"SPECIFY ORDER OF EACH SUBSYSTEM"/1X,"5I2")
READ(IRD,111) (II(K),K=1,IS)
DO 800 K=1,IS
800 WRITE(IW,5545) K,II(K)
5545 FORMAT(1X,"ORDER OF SUBSYSTEM",I2,"=",I2)
C
C START DECOMPOSITION
C
C 215 CONTINUE
C
C DECOMPOSE SYSTEM INTO SUBSYSTEMS
C
C CALL DECP(IS,M,N)
C
C START STABILIZATION
C
C DO 10 K=1,IS
C L1=II(K)
C IP=(K-1)*IS+K
C DO 20 L=1,L1
C DO 20 J=1,L1
20 AK(L,J)=A2(IP,L,J)
C DO 30 L=1,L1
30 B3(L)=B2(K,L,K)
C
C LOCATE POLES OF EACH SUBSYSTEM
C
C CALL PPL(L1,IW)
C DO 15 L=1,L1
15 T1(L,K)=R1(L)
10 CONTINUE
WRITE(IW,500)
500 FORMAT(1X,"DO YOU WANT TO CONTINUE,YES OR NO")
READ(1,501) A10
501 FORMAT(A2)
IF(A10,NE,DD) GO TO 266
C
C START ITERATION FOR ALFA PARAMETER
C
C WRITE(IW,600)
600 FORMAT(1X,"SPECIFY INCREME DELT"/1X,"F10.0")
READ(1,200) DELT
200 FORMAT(F10.0)
ALF=1.
211 CONTINUE
C
C FORM VANDERMONDE MATRIX
C
C DO 112 K=1,IS
C L1=II(K)

```

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```

DO 112 J=1,L1
SK(J,K)=T1(J,K)
112 SK(J,K)=ALF*SK(J,K)
DO 12 K=1,15
L1=II(K)
DO 88 L=1,L1
DO 88 J=1,LL
88 Q(K,L,J)=SK(J,K)**(L-1)
12 CONTINUE

```

```

C
C C
C FORM AGGREGATE MODEL

```

```

CALL DECP(IS,M,N)
CALL TRF(IS)
CALL AGR(IS)
WRITE(IW,531)
531 FORMAT(1X,"AGGREGATION MATRIX")
DO 281 K=i,IS
281 WRITE(6,105) (Q1(K,J),J=1,IS)
ALF=ALF*DELT

```

```

C
C C
C CHECK THE SIGN OF THE K-TH MINOR

```

```

IZ=1
DO 282 K=1,15.
IF=K
DO 283 I=1,IF
DO 283 J=1,IF
IP=(I-1)*K+J
283 C(IP)=Q1(J,I)
CALL MINV(C,IF,D,LL,MM)
IZ=-IZ
D=IZ*D
WRITE(6,270)D
270 FORMAT(1X,F14.5)
IF(D)211,211,282

```

```

C
C C
C PRINT CORRESPONDING EIGENVALUES

```

```

282 CONTINUE
DO 11 K=1,15
L1=II(K)
WRITE(IW,700)K
700 FORMAT(1X,"EIGENVALUES OF SUBSYSTEM",I2)
DO 18 J=1,L1
18 WRITE(6,701)J,SK(J,K)
701 FORMAT(1X,"LAMBDA(",I2,") = ",F15.6)
11 CONTINUE
GO TO 215
266 CONTINUE
END
ENDS

```

```
0161 SUBROUTINE BIG1(R1,L,N)
0162 C
0163 C PROGRAM TO CALCULATE THE LARGEST ELEMENT OF AN ARRAY
0164 C
0165 DIMENSION R1(6)
0166 K=1
0167 J=K+1
0168 10 IF(R1(K).GE.R1(J))GO TO 8
0169 K=J
0170 8 J=J+1
0171 IF(J.LE.N)GO TO 10
0172 L=K
0173 RETURN
0174 END
```

```
** NO ERRORS** PROGRAM = 00055 COMMON = 00000
```

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```
0175      SUBROUTINE TRAN1(Q1,AG,LI,LS)
0176      C
0177      C PROGRAM TO TRANSPOZE A MATRIX
0178      C
0179      DIMENSION Q1(5,5),AG(5,5)
0180      DO 8 K=1,LI
0181      DO 8 J=1,LS
0182      8 AG(J,K)=Q1(K,J)
0183      RETURN
0184      END
```

\*\* NO ERRORS\*\* PROGRAM = 00056 COMMON = 00000

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```
0185      SUBROUTINE SMALL(R1,L,N)
0186      C
0187      C      PROGRAM TO CALCULATE THE SMALLEST ELEMENT OF AN ARRAY
0188      C
0189      DIMENSION R1(6)
0190      K=1
0191      J=K+1
0192      10 IF(R1(K).LE.R1(J))GO TO 8
0193      9 K=J
0194      8 J=J+1
0195      IF(J.LE.N)GO TO 10
0196      L=K
0197      RETURN
0198      END
```

```
** NO ERRORS**      PROGRAM = 00055      COMMON = 00000
```

```

0199      SUBROUTINE KBAR(A,N,Z,IZ,D,B)
0200      C
0201      C PROGRAM TO CALCULATE GAIN VECTOR
0202      C
0203      DIMENSION A(5*5),Z(1),D(6),B(5),Q(5*5),R(5),P1(25),LL(5),MM(5)
0204      1,C(5)
0205      CALL COEF1(A,N,D)
0206      DO 8 K=1,N
0207      8 Z(K)=D(K)-Z(K)
0208      DO 9 I=1,N
0209      9 Q(I,N)=B(I)
0210      L=N-1
0211      DO 99 K=1,L
0212      LI=N-K+1
0213      LZ=N-K
0214      F=D(LI)
0215      DO 7 I=1,N
0216      7 B(I)=Q(I,N)
0217      CALL SCALU(B,N,F)
0218      DO 77 I=1,N
0219      77 R(I)=Q(I,LI)
0220      CALL VECPR(A,R,N,C)
0221      DO 10 I=1,N
0222      10 Q(I,LZ)=C(I)+B(I)
0223      99 CONTINUE
0224      DO 200 K=1,N
0225      DO 200 J=1,N
0226      I=(K-1)*N+J
0227      200 P1(I)=Q(J,K)
0228      DO 105 K=1,N
0229      105 B(K)=Q(K,N)
0230      CALL MINV(P1,N,SD,LL,MM)
0231      IF(SD)5,4,5
0232      4 WRITE(6,102)
0233      102 FORMAT(1X,13HSINGULAR CASE)
0234      5 CONTINUE
0235      DO 201 K=1,N
0236      DO 201 J=1,N
0237      I=(K-1)*N+J
0238      201 Q(J,K)=P1(I)
0239      DO 91 K=1,N
0240      S=0.
0241      DO 92 J=1,N
0242      92 S=S+Z(J)*Q(J,K)
0243      91 D(K)=S
0244      RETURN
0245      END

```

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\*\* NO ERRORS\*\* PROGRAM = 00545 COMMON = 00000

```
0246      SUBROUTINE DECP(IS,M,N)
0247      C
0248      C      PROGRAM TO DECOMPOSE SYSTEM INTO SUBSYSTEMS
0249      C
0250      DIMENSION A(10,10),B(10,3),A2(9,10,10),B2(3,10,3),II(5)
0251      COMMON A,B,A2,B2,II
0252      IL=1
0253      L1=0
0254      DO 10 K=1,IS
0255      IP=1
0256      L11=II(K)
0257      L1=L1+L11
0258      L2=0
0259      DO 11 J=1,IS
0260      NN=(K-1)*IS+J
0261      L22=II(J)
0262      L2=L2+L22
0263      IR=0
0264      DO 12 JJ=IL,L1
0265      IR=IR+1
0266      IC=0
0267      DO 12 KK=IP,L2
0268      IC=IC+1
0269      12 A2(NN,IR,IC)=A(JJ,KK)
0270      11 IP=IP+L22
0271      IK=0
0272      DO 13 JJ=IL,L1
0273      IR=IR+1
0274      DO 13 KK=1,M
0275      13 B2(K,IR,KK)=B(JJ,KK)
0276      IL=IL+L11
0277      10 CONTINUE
0278      RETURN
0279      END
```

```
** NO ERRORS**      PROGRAM = 00187      COMMON = 02245
```

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```

0280      SUBROUTINE PPL(N,IW)
0281      C
0282      C      POLE SHIFTING PROGRAM
0283      C
0284      INTEGER DD
0285      INTEGER YES
0286      REAL K1
0287      DIMENSION A1(10,10),B1(10,3),A2(9,10,10),B2(3,10,3),I1(5),A(5,5),
0288      IQ1(5,5),Q(3,5,5),I11(5,5),B(5),R1(6),R2(6),D(6),SK(6,10),COF(6)
0289      I,Z(B)
0290      COMMON A1,B1,A2,B2,I1,A,Q1,Q,I11,B,R1,R2,D,SK
0291      DATA YES/2HYES/
0292      WRITE(IW,153)
0293      153 FORMAT(15X,"POLES OF THE SYSTEM")
0294      CALL ALAM(A,N,D,COF,R1,R2)
0295      DO 19 K=1,N
0296      19 WRITE(IW,100)K,R1(K),R2(K)
0297      100 FORMAT(1X,8HLAMBDA (,I1,4H) =,F14.6,4H + J,F14.6)
0298      WRITE(IW,200)
0299      200 FORMAT(1X,"DO YOU WANT TO ALTER THE POLES ?,YES OR NO?")
0300      READ(I,330)DD
0301      330 FORMAT(AZ)
0302      IF(YES.NE.DD)GO TO 340
0303      WRITE(IW,500)
0304      500 FORMAT(1X,"ENTER DESIRED EIGENVALUES")
0305      WRITE(IW,510)
0306      510 FORMAT(1X,"ASSIGN INPUT UNIT"/1X,"I2")
0307      READ(I,111)IRD
0308      111 FORMAT(I2)
0309      DO 209 J=1,N
0310      WRITE(IW,501)
0311      501 FORMAT(1X,"      ",",",",")
0312      209 READ(IRD,208)R1(J),R2(J)
0313      208 FORMAT(2F14.6)
0314      CALL DISP(R1,R2,N,Z)
0315      CALL KBAR(A,N,Z,IZ,D,B)
0316      DO B I=1,N
0317      DO B J=1,N
0318      B A(I,J)=A(I,J)+B(I)*D(J)
0319      WRITE(IW,919)
0320      919 FORMAT(20X,8HVECTOR K//)
0321      WRITE(IW,109) (D(K),K=1,N)
0322      109 FORMAT(1X,5F14.6)
0323      340 CONTINUE
0324      RETURN
0325      END

```

\*\* NO ERRORS\*\*

PROGRAM = 00434

COMMON = 02686

```

0326     SUBROUTINE AGR(IS)
0327 C
0328 C PROGRAM TO FORM AN AGGREGATE MATRIX
0329 C
0330     REAL KB
0331     DIMENSION A(10,10),B(10,3),A2(9,10,10),B2(3,10,3),II(5),A4(5,5)
0332     ,A3(5,5),Q(3,5,5),I11(5,5),B3(5),R1(6),R2(6),KB(6),SK(6,10),
0333     ICOF(6),D(6),Q1(5,5)
0334     COMMON A,B,A2,B2,II,A4,A3,Q,I11,B3,R1,R2,KB,SK
0335     DO 5 K=1,IS
0336     L1=II(K)
0337     DO 5 J=1,IS
0338     L2=II(J)
0339     IT=(K-1)*IS+J
0340     IF(K.EQ.J)GO TO 13
0341     DO 8 L=1,L1
0342     DO 8 I=1,L2
0343     8 A*(I,L)=A2(IT,L,I)
0344     DO 9 L=1,L2
0345     DO 9 I=1,L2
0346     S=0.
0347     DO 10 IR=1,L1
0348     10 S=S+A4(I,IR)*A2(IT,IR,L)
0349     9 Q1(I,L)=S
0350     CALL ALAM(Q1,L2,D,COF,R2,KB)
0351     DO 11 I=1,L2
0352     IF(R2(I).LE.0.)GO TO 11
0353     R2(I)=SQRT(R2(I))
0354     11 CONTINUE
0355     CALL BIG1(R2,L,L2)
0356     GO TO 14
0357     13 CONTINUE
0358     DO 20 I=1,L1
0359     20 R2(I)=-SK(I,K)
0360     CALL SMAL1(R2,L,L1)
0361     R2(L)=-R2(L)
0362     14 A3(K,J)=R2(L)
0363     5 CONTINUE
0364     RETURN
0365     END

```

\*\* NO ERRORS\*\* PROGRAM = 00383 COMMON = 02686

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```

0366      SUBROUTINE TRF(IS)
0367      C
0368      C      PROGRAM TO TRANSFORM SUBSYSTEM BY SIMILARITY TRANSFORMATION
0369      C
0370      DIMENSION A(10,10),B(10,3),A2(9,10,10),B2(3,10,3),II(5),A1(5,5)
0371      I,Q1(5,5),Q(3,5,5),I11(5),B3(5)
0372      I,C(25),LL(5),MM(5)
0373      COMMON A,B,A2,B2,II,A1,Q1,Q,I11,B3
0374      DO 25 K=1,IS
0375      L1=II(K)
0376      DO 26 J=1,L1
0377      DO 26 L=1,L1
0378      IP=(J-1)*L1+L
0379      26 C(IP)=Q(K,L,J)
0380      CALL MINV(C,L1,D,LL,MM)
0381      DO 27 J=1,L1
0382      DO 27 L=1,L1
0383      IP=(J-1)*L1+L
0384      27 Q1(L,J)=C(IP)
0385      DO 33 JJ=1,IS
0386      IT=(K-1)*IS+JJ
0387      L2=II(JJ)
0388      DO 28 J=1,L1
0389      DO 28 L=1,L2
0390      S=0.
0391      DO 29 IZ=1,L1
0392      29 S=S+Q1(J,IZ)*A2(IT,IZ,L)
0393      28 A1(J,L)=S
0394      DO 31 J=1,L1
0395      DO 31 L=1,L2
0396      S=0.
0397      DO 32 IZ=1,L2
0398      32 S=S+A1(J,IZ)*Q(JJ,IZ,L)
0399      31 A2(IT,J,L)=S
0400      33 CONTINUE
0401      DO 39 J=1,L1
0402      S=0.
0403      DO 34 L=1,L1
0404      34 S=S+Q1(J,L)*B2(K,L,K)
0405      39 B3(J)=S
0406      DO 20 J=1,L1
0407      20 B2(K,J,K)=B3(J)
0408      25 CONTINUE
0409      DO 72 K=1,IS
0410      L1=II(K)
0411      DO 73 J=1,L1
0412      DO 73 L=1,L1
0413      IP=(J-1)*L1+L
0414      73 C(IP)=Q(K,L,J)
0415      CALL MINV(C,L1,D,LL,MM)
0416      DO 74 J=1,L1
0417      DO 74 L=1,L1
0418      IP=(J-1)*L1+L
0419      74 Q(K,L,J)=C(IP)
0420      72 CONTINUE
0421      RETURN

```

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0422 END

\*\* NO ERRORS\*\* PROGRAM = 00561 COMMON = 02510

```
0423     SUBROUTINE ALAM(A,N,D,COF,R1,R2)
0424     C
0425     C     PROGRAM TO CALCULATE EIGENVALUES OF GENERAL N BY N MATRIX
0426     C
0427     DIMENSION A(5,5),D(6),COF(6),R1(6),R2(6)
0428     CALL CDEF1(A,N,D)
0429     M1=N+1
0430     CALL POLRT(D,COF,N,R1,R2,IER)
0431     IF(IER.EQ.0)GO TO 10
0432     WRITE(6,100)IER
0433 100  FORMAT(1X,13HERROR CODE = ,I2)
0434     GO TO 10
0435     RETURN
0436     END
```

\*\* NO ERRORS\*\* PROGRAM = 00064 COMMON = 00000

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```
0437      SUBROUTINE UNIT1(R,N).
0438 C
0439 C      SUBROUTINE TO FORM UNIT MATRIX
0440 C
0441      DIMENSION R(5,5)
0442      DO 9 K=1,N
0443      DO 9 J=1,N
0444      9 R(K+J)=0.
0445      DO 8 K=1,N
0446      8 R(K+K)=1.
0447      RETURN
0448      END
```

```
** NO ERRORS**      PROGRAM = 00072      COMMON = 00000
```

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```
0449      SUBROUTINE COEF1(A,N,D)
0450      C C
0451      C C      SUBROUTINE TO CALCULATE COEFFICIENTS OF CHARACTERISTIC POLYNOMIAL
0452      C C      OF MATRIX A
0453      C
0454      DIMENSION A(5,5),B(5,5),C(5,5),D(6)
0455      CALL UNIT1(B,N)
0456      DO 9 K=1,N
0457      CALL PROD1(A,B,N,N,N)
0458      CALL TRAC1(B,N,S)
0459      L=N-K+1
0460      D(L)=-1./K)*S
0461      CALL UNIT1(C,N)
0462      DI=D(L)
0463      CALL SCH1(C,D1,N)
0464      9 CALL ADD1(C,B,N)
0465      L=N+1
0466      D(L)=1.
0467      RETURN
0468      END
```

```
** NO ERRORS**      PROGRAM = 00211      COMMON = 00000
```

```
0469      SUBROUTINE SCALU(B,N,F)
0470      C
0471      C      SUBROUTINE TO MULTIPLY MATRIX BY A SCALAR
0472      C
0473      DIMENSION B(5)
0474      DO B K=1,N
0475      B(K)=F*B(K)
0476      RETURN
0477      END
```

\*\* NO ERRORS\*\* PROGRAM = 00041 COMMON = 00000

```
0478      SUBROUTINE VECPR(A,B,N,C)
0479 C
0480 C      PROGRAM TO MULTIPLY MATRIX BY A VECTOR
0481 C
0482      DIMENSION A(S,S),B(S),C(S)
0483      DO 9 K=1,N
0484      S=0.
0485      DO 8 J=1,N
0486      B S=S+A(K,J)*B(J)
0487      9 C(K)=S
0488      RETURN
0489      END
```

\*\* NO ERRORS\*\* PROGRAM = 00075 COMMON = 00000

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```
0490      SUBROUTINE DISP(R1,R2,N,Z)
0491      CCC
0492      FORM POLYNOMIAL FROM ITS ZEROS
0493
0494      DIMENSION R1(1),R2(1),Z(1),X(8),Y(8)
0495      K=1
0496      Y(1)=1.
0497      Y(2)=0.
0498      IY=2
0499      20 IF(R2(K).EQ.0.)GO TO 10
0500      X(1)=R1(K)**2+R2(K)**2
0501      X(2)=-2.*R1(K)
0502      X(3)=1.
0503      IX=3
0504      K=K+1
0505      GO TO 11
0506      10 X(1)=-R1(K)
0507      X(2)=1.
0508      IX=2
0509      11 CALL PMPY(Z,IZ,X,IX,Y,IY)
0510      DO 8 L=1,IZ
0511      8 Y(L)=Z(L)
0512      IY=IZ
0513      K=K+1
0514      IF(K.LE.N)GO TO 20
0515      RETURN
0516      END
```

```
** NO ERRORS**      PROGRAM = 00230      COMMON = 00000
```

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```
0517      SUBROUTINE PROD1(A,B,N,I,M)
0518      C
0519      C      SUBROUTINE TO MULTIPLY TWO MATRICES
0520      C
0521      DIMENSION A(5,5),B(5,5),C(5,5)
0522      S=0.
0523      DO 9 K=1,N
0524      DO 9 L=1,M
0525      S=0.
0526      DO 10 J=1,I
0527      10 S=S+A(K,J)*B(J,L)
0528      9 C(K,L)=S
0529      DO 8 K=1,N
0530      DO 8 L=1,M
0531      8 B(K,L)=C(K,L)
0532      RETURN
0533      END
```

```
** NO ERRORS**      PROGRAM = 00183      COMMON = 00000
```

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```
0534 SUBROUTINE ADD1(A,B,N)
0535 C
0536 C SUBROUTINE TO ADD TWO MATRICES
0537 C
0538 DIMENSION A(5,5),B(5,5)
0539 DO 8 K=1,N
0540 DO 8 J=1,N
0541 B(K,J)=B(K,J)+A(K,J)
0542 RETURN
0543 END
```

\*\* NO ERRORS\*\* PROGRAM = 00066 COMMON = 00000

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```
0544      SUBROUTINE TRAC1(A,N,S)
0545      C
0546      C      SUBROUTINE TO CALCULATE TRACE OF A MATRIX
0547      C
0548      DIMENSION A(5,5)
0549      S=0.
0550      DO 8 K=1,N
0551      8 S=S+A(K,K)
0552      RETURN
0553      END
```

\*\* NO ERRORS\*\* PROGRAM = 00044 COMMON = 00000



```
0554      SUBROUTINE SCH1(A,D,N)
0555      C
0556      C      SUBROUTINE TO MULTIPLY MATRIX BY A SCALAR
0557      C
0558      DIMENSION A(5,5)
0559      DO 9 K=1,N
0560      DO 9 J=1,N
0561      9 A(K,J)=D*A(K,J)
0562      RETURN
0563      END
```

\*\* NO ERRORS\*\* PROGRAM = 00057 COMMON \* 00000

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0564

ENDS

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## A.2. Optimization Program

The only step that may introduce some computational complexities in the optimization scheme described in Section 4 is the solution of the matrix Riccati equation for the evaluation of the local controls. Despite the fact that this computation is performed at the subsystem level and hence involves matrices of small orders, simulation on a digital computer will invariably be necessary. Although many different methods for the solution of the Riccati equation exist in the literature, the particular method that is adopted here is the iterative technique due to Kleinman [19]. In addition to determining the symmetric positive-definite solution  $P$  of the Riccati equation

$$A^T P + PA - PBR^{-1}B^T P + Q = 0 ,$$

the program described here also computes the eigenvalues of the matrices  $P$  and  $W = Q + PBR^{-1}B^T P$  that is necessary in the evaluation of the suboptimality index  $\epsilon$ .

The simulation analysis was conducted on the HP 2100 digital computing system (32K memory) in FORTRAN language. In the following description, only the subroutines MINV, SIMQ and POLRT are to be supplied externally (from IBM Scientific Subroutine Package), while the rest are contained internally. Since the computation involves only the subsystems that result from a suitable decomposition of the overall system and hence are necessarily of small dimensions, the program is prepared to handle subsystems of dimension up to five.

### *DESCRIPTION OF THE EXTERNAL SUBROUTINES (From SSP)*

#### Subroutine MINV

##### *Purpose:*

Invert a matrix.

*Usage:*

CALL MINV (A, N, D, L, M)

*Description of parameters:*

- A - Input matrix, destroyed in computation and replaced by resultant inverse.
- N - Order of matrix A.
- D - Resultant determinant.
- L - Work vector of length N.
- M - Work vector of length N.

*Remarks:*

Matrix A must be a general (nonsingular) matrix.

*Subroutines and function subprograms required:* None.

*Method:*

The standard Gauss-Jordan method is used. The determinant is also calculated. A determinant with absolute value less than  $10^{**}(-20)$  indicates singularity.

Subroutine SIMQ*Purpose:*

Obtain solution of a set of simultaneous linear equations  $AX = b$ .

*Usage:*

CALL SIMQ (A, B, N, KS)

*Description of parameters:*

- A - Matrix of coefficients stored columnwise. These are destroyed in the computation. The size of matrix A is N by N.
- B - Vector of original constants (length N). These are replaced by final solution values, vector X.

N - Number of equations and variables. N must be greater than 1.

KS - Output digit: 0 for a normal solution; 1 for a singular set of equations.

*Remarks:*

Matrix A must be general. If matrix is singular, solution values are meaningless.

*Subroutines and function subprograms required:* None.

*Method:*

Method of solution is by elimination using largest pivotal divisor.

Subroutine POLRT

*Purpose:*

Computes the real and complex roots of a real polynomial.

*Usage:*

CALL POLRT (XCOF, COF, M, ROOTR, ROOTI, IER)

*Description of parameters:*

XCOF - Vector of M+1 coefficients of the polynomial ordered from smallest to largest power.

COF - Working vector of length M+1.

M - Order of polynomial.

ROOTR - Resultant vector of length M containing real roots of the polynomial.

ROOTI - Resultant vector of length M containing the corresponding imaginary roots of the polynomial.

IER - Error code where

IER = 0 No error

IER = 1 M less than one

IER = 2 M greater than 36

IER = 3 Unable to determine root with 500  
iterations on 5 starting values.

IER = 4 High order coefficient is zero.

*Remarks:*

Limited to 36-th order polynomial or less. Floating point overflow may occur for high order polynomials but will not affect the accuracy of the results.

*Subroutines and function subprograms required:* None

*Method:*

Newton-Raphson iterative technique.

```

0001 FIN4:1
0002 PROGRAM RICAT
0003 C SOLUTION OF ALGEBRAIC RICCATI EQUATION BY KLEINMAN ITERATION
0004 DIMENSION A(5,5),B(5,5),Q(5,5),R(5,5),S0(5,5),S1(5,5),
0005 1B5(5,5),AM(5,5),TS(5,5),AUX(5,5),D(5,5),P(5,5),W(5,5),
0006 2R1N(5,5),SF(5,5),LV(5,5),MV(5,5),VR(25),TB(5,5),TA(5,5),
0007 3AJAA(5,5),AUXB(5,5),AUXC(5,5),AUXD(5,5),AUXE(5,5),AUXF(5,5),
0008 4AJAG(5,5),AUXH(5,5),VD(6),COF(6),R1(6),R2(6)
0009 C READ THE STATE DIMENSION N
0010 READ(5,1)N
0011 1 FORMAT(15)
0012 C READ THE CONTROL DIMENSION M
0013 READ(5,1)M
0014 C READ THE SYSTEM MATRIX A
0015 READ(5,2)((A(I,J),J=1,N),I=1,N)
0016 2 FORMAT(6F10.5)
0017 C READ THE STATE-CONTROL MATRIX B
0018 READ(5,2)((B(I,J),J=1,M),I=1,N)
0019 C READ THE COST MATRICES Q AND R
0020 READ(5,2)((Q(I,J),J=1,N),I=1,N)
0021 READ(5,2)((R(I,J),J=1,M),I=1,M)
0022 WRITE(6,20)
0023 20 FORMAT(3X,"SOLUTION OF RICCATI EQUATION-KLEINMAN ITERATIVE METHOD")
0024 1,////)
0025 WRITE(6,21)
0026 21 FORMAT(5X,"SYSTEM MATRIX A",//)
0027 DO 22 I=1,N
0028 22 WRITE(6,23) (A(I,J),J=1,N)
0029 23 FORMAT(140,10(2X,E13.5))
0030 WRITE(6,24)
0031 24 FORMAT(///)
0032 WRITE(6,25)
0033 25 FORMAT(5X,"STATE-CONTROL MATRIX B",//)
0034 DO 25 I=1,N
0035 25 WRITE(6,23) (B(I,K),K=1,M)
0036 WRITE(6,24)
0037 WRITE(6,27)
0038 27 FORMAT(5X,"COST MATRICES Q AND R",//)
0039 WRITE(6,28)
0040 28 FORMAT(140,"Q MATRIX",//)
0041 DO 29 I=1,N
0042 29 WRITE(6,23) (Q(I,J),J=1,N)
0043 WRITE(6,30)
0044 30 FORMAT(//,1X,"R MATRIX",//)
0045 DO 31 I=1,M
0046 31 WRITE(6,23) (R(I,J),J=1,M)
0047 WRITE(6,24)
0048 C READ THE STARTING APPROXIMANT S0
0049 READ(5,2)((S0(I,J),J=1,N),I=1,M)
0050 DO 3 I=1,M
0051 DO 3 J=1,N
0052 3 S1(I,J)=S0(I,J)
0053 CALL MOVE(R,M,VR)
0054 CALL MINV(VR,M,VD,LY,MV)
0055 CALL VEM(VR,M,MIN)
0056 C FIRST ITERATION COMMENCES HERE

```

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```

0057 32 WRITE(5,33)
0058 33 FORMAT(10X,"START OF THE FIRST ITERATION",///)
0059 WRITE(5,34)
0060 34 FORMAT(5X,"STARTING APPROXIMANT-S MATRIX",///)
0061 DO 35 I=1,M
0062 35 WRITE(6,23) (S(I,J),J=1,N)
0063 WRITE(5,24)
0064 DO 200 NVAR=1,50
0065 CALL MAMUL(B,S,B5,N,M,N)
0066 DO 4 I=1,N
0067 DO 4 J=1,N
0068 4 B5(I,J)=-B5(I,J)
0069 C THE MODIFIED SYSTEM MATRIX AM IS OBTAINED NOW
0070 CALL MASUM(A,B5,AM,N,N)
0071 WRITE(5,36)
0072 36 FORMAT(5X,"MODIFIED SYSTEM MATRIX--AM=A-B*S",///)
0073 DO 37 I=1,N
0074 37 WRITE(5,23) (AM(I,J),J=1,N)
0075 WRITE(6,24)
0076 141 CALL EIVAL(AM,N,VD,COF,R1,R2)
0077 WRITE(5,142)
0078 142 FORMAT(5X,"EIGENVALUES OF CLOSED LOOP MATRIX",///)
0079 DO 143 J=1,N
0080 143 WRITE(5,23) R1(J),R2(J)
0081 WRITE(5,24)
0082 C THE LYAPUNOV MATRIX EQUATION IS NOW FORMED
0083 CALL MATRN(S,T5,M,N)
0084 CALL MAMUL(T5,R,AUX0,N,M,M)
0085 CALL MAMUL(AUX0,S,AUXH,N,M,N)
0086 CALL MASUM(Q,AUXH,D,N,N)
0087 C THE LYAPUNOV MATRIX EQUATION IS NOW P*AM*TR(AM)*P*-D
0088 CALL LYAPU(D,AM,P,N)
0089 7 WRITE(5,8)
0090 8 FORMAT(5X,"P MATRIX-SOLUTION OF LYAPUNOV EQUATION",///)
0091 DO 9 I=1,N
0092 9 WRITE(6,10) (P(I,J),J=1,N)
0093 10 FORMAT(140,10(2X,E13.5))
0094 WRITE(6,24)
0095 C COMPUTE THE APPROXIMANT FOR THE NEXT ITERATION
0096 CALL MATRN(B,TB,N,M)
0097 CALL MAMUL(RIN,TB,AUX,M,M,N)
0098 CALL MAMUL(AUX,P,SF,M,N,N)
0099 CALL MATRN(A,TB,N,N)
0100 CALL MAMUL(TA,P,AUXA,N,N,N)
0101 CALL MAMUL(P,A,AUXB,N,N,N)
0102 CALL MAMUL(B,SF,AUXC,N,M,N)
0103 CALL MAMUL(P,AUXC,AUX,N,N,N)
0104 C COMPUTE THE W MATRIX, W=Q+P*B*RIN*TB*P
0105 CALL MASUM(Q,AUX,W,N,N)
0106 WRITE(5,131)
0107 131 FORMAT(5X,"W MATRIX --- W=Q+P*B*RIN*TB*P",///)
0108 DO 132 I=1,N
0109 132 WRITE(5,23) (W(I,J),J=1,N)
0110 WRITE(6,24)
0111 DO 112 I=1,N
0112 DO 112 J=1,N

```

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```

0113 112 AUX(I,J)=-AUX(I,J)
0114 CALL MASUM(AUX,AUXA,AUXF,N,N)
0115 CALL MASUM(AUXF,AUXU,AUXD,N,N)
0116 CALL MASUM(AUXD,Q,AUXE,N,N)
0117 C CHECKING THE SOLUTION
0118 WRITE(6,111)
0119 111 FORMAT(10X,"CHECKING THE SOLUTION",//)
0120 113 WRITE(6,114)
0121 114 FORMAT(5X,"VALUE OF RICCATI MATRIX",//)
0122 DO 115 I=1,N
0123 115 WRITE(6,23)(AUXE(I,J),J=1,N)
0124 WRITE(6,24)
0125 C MATRIX SF IS THE STARTING APPROXIMANT FOR NEXT ITERATION
0126 C THE STEADY VALUE OF SF IS THE FEEDBACK GAIN MATRIX FOR THE OPTIMAL
0127 C REGULATOR K=N14*10*P
0128 DO 129 I=1,M
0129 DO 129 J=1,N
0130 S(I,J)=5F(I,J)
0131 C ADDITIONAL COMPUTATIONS--CALCULATION OF EIGENVALUES OF
0132 C MATRICES W AND P--THIS PART MAY BE OMITTED WHEN THE SOLUTION
0133 C OF A RICCATI EQUATION ONLY IS DESIRED
0134 CALL EIVAL(P,N,VD,COF,R1,R2)
0135 WRITE(5,133)
0136 133 FORMAT(5X,"EIGENVALUES OF P",//)
0137 DO 134 J=1,N
0138 134 WRITE(6,23)R1(J),R2(J)
0139 WRITE(6,24)
0140 CALL EIVAL(W,N,VD,COF,R1,R2)
0141 WRITE(6,135)
0142 135 FORMAT(5X,"EIGENVALUES OF W",//)
0143 DO 136 I=1,N
0144 136 WRITE(6,23)R1(I),R2(I)
0145 WRITE(6,24)
0146 WRITE(5,38)
0147 38 FORMAT(10X,"NEXT ITERATION COMMENCES HERE",//)
0148 WRITE(6,106)
0149 106 FORMAT(5X,"MODIFIED APPROXIMANT-NEW S MATRIX",//)
0150 DO 39 J=1,M
0151 39 WRITE(6,23)(S(I,J),J=1,N)
0152 WRITE(6,24)
0153 200 CONTINUE
0154 99 STOP
0155 END

```

\*\* NO ERRORS\*\* PROGRAM = 02921 COMMON = 00000

PAGE 0001

FTN4 COMPILER: HP24177 (SEPT. 1974)

```
0156 SUBROUTINE MAHUL(F,G,H,N1,N2,N3)
0157 DIMENSION F(5,5),G(5,5),H(5,5)
0158 DO 51 I=1,N1
0159 DO 51 J=1,N3
0160 SUM=0.0
0161 DO 52 K=1,N2
0162 52 SUM=SUM+F(I,K)*G(K,J)
0163 H(I,J)=SUM
0164 51 CONTINUE
0165 RETURN
0166 END
```

\*\* NO ERRORS\*\* PROGRAM = 00091 COMMON = 00000

PAGE 0001

FTN4 COMPILER: HP24177 (SEPT. 1974)

```
0167 SUBROUTINE MASUM(F,G,H,N1,N2)
0168 DIMENSION F(5,5),G(5,5),H(5,5)
0169 DO 61 I=1,N1
0170 DO 61 J=1,N2
0171 H(I,J)=F(I,J)+G(I,J)
0172 61 CONTINUE
0173 RETURN
0174 END
```

\*\* NO ERRORS\*\* PROGRAM = 00067 COMMON = 00000

PAGE 0001

FTN4 COMPILER: HP24177 (SEPT. 1974)

```
0175 SUBROUTINE MATRN(F,G,N1,N2)
0176 DIMENSION F(5,5),G(5,5)
0177 DO 71 I=1,N1
0178 DO 71 J=1,N2
0179 G(J,I)=F(I,J)
0180 71 CONTINUE
0181 RETURN
0182 END
```

\*\* NO ERRORS\*\* PROGRAM = 00055 COMMON = 00000

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PAGE 0001

FTN4 COMPILER: HP24177 (SEPT. 1974)

```
0183 SUBROUTINE LYAPU(G,A,H,N)
0184 DIMENSION G(5,5),A(5,5),H(5,5),P(15),L(5,5),U(15,15),P1(225)
0185 M=(N+1)*N/2
0186 CALL TRIN(G,P,N,M)
0187 CALL MATL(N,H,L)
0188 CALL LP(L,A,N,M,U)
0189 DO 10 I=1,M
0190 10 P(I)=-P(I)
0191 CALL MAVEC(U,H,P1)
0192 CALL SIMQ(P1,P,M,K5)
0193 IF(K5=1)11,12,11
0194 12 WRITE(6,100)
0195 100 FORMAT(1X,13HSINGULAR CASE)
0196 11 CALL ATRI(P,H,M,N)
0197 RETURN
0198 END
```

\*\* NO ERRORS\*\* PROGRAM = 01076 COMMON \* 00000

PAGE 0001

FTN4 COMPILER: HP24177 (SEPT. 1974)

```
0199 SUBROUTINE TRIN(A,P,N,M)
0200 DIMENSION A(5,5),P(15)
0201 M=0
0202 DO 8 K=1,N
0203 L=K
0204 DO 8 I=L,N
0205 H=M+1
0206 8 P(M)=A(L,I)
0207 RETURN
0208 END
```

\*\* NO ERRORS\*\* PROGRAM \* 00064 COMMON \* 00000

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PAGE 0001

FTN4 COMPILER: HP24177 (SEPT. 1974)

```
0209      SUBROUTINE ATRI(P,A,M,N)
0210      DIMENSION A(5,5),P(15)
0211      M=0
0212      DO 8 K=1,N
0213      L=K
0214      DO 8 I=L,N
0215      M=M+1
0216      8 A(L,I)=P(M)
0217      DO 9 J=1,N
0218      L=J
0219      DO 9 I=L,N
0220      9 A(I,J)=A(J,I)
0221      RETURN
0222      END
```

\*\* NO ERRORS\*\* PROGRAM = 00104 COMMON = 00000

PAGE 0001

FTN4 COMPILER: HP24177 (SEPT. 1974)

```
0223      SUBROUTINE MATL(N,M,L)
0224      DIMENSION L(5,5),LI(15)
0225      DO 6 K=1,M
0226      8 LI(K)=K
0227      MI=0
0228      DO 18 K=1,N
0229      LS=K
0230      DO 18 I=LS,N
0231      MI=MI+1
0232      18 L(LS,I)=LI(MI)
0233      DO 9 J=1,N
0234      LS=J
0235      DO 9 I=LS,N
0236      9 L(I,J)=L(J,I)
0237      RETURN
0238      END
```

\*\* NO ERRORS\*\* PROGRAM = 00128 COMMON = 00000

PAGE 0001

FTN4 COMPILER: HP24177 (SEPT. 1974)

```
0239      SUBROUTINE MAVEC(A,N,P)
0240      DIMENSION A(15,15),P(225)
0241      DO 6 K=1,N
0242      DO 8 J=1,N
0243      L=N*(K-1)+J
0244      8 P(L)=A(J,K)
0245      RETURN
0246      END
```

\*\* NO ERRORS\*\* PROGRAM = 00061 COMMON = 00000

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```

0247      SUBROUTINE MAVR(A,N,P1)
0248      DIMENSION A(5,5),P1(25)
0249      DO 8 K=1,N
0250      DO 8 J=1,N
0251      L=N*(K-1)+J
0252      B P1(L)=A(J,K)
0253      RETURN
0254      END

```

\*\* NO ERRORS\*\*      PROGRAM = 00061      COMMON = 00000

```

0255      SUBROUTINE VEMA(P1,N,A)
0256      DIMENSION A(5,5),P1(25)
0257      DO 8 K=1,N
0258      DO 8 J=1,N
0259      L=N*(K-1)+J
0260      B A(J,K)=P1(L)
0261      RETURN
0262      END

```

\*\* NO ERRORS\*\*      PROGRAM = 00062      COMMON = 00000

```

0263      SUBROUTINE LP(L,A,N,M,U)
0264      DIMENSION L(5,5),A(5,5),U(15,15)
0265      DO 11 I=1,M
0266      DO 11 J=1,M
0267      11 U(I,J)=0.
0268      DO 12 I=1,N
0269      DO 12 J=1,N
0270      DO 12 K=1,N
0271      II=L(I,K)
0272      IJ=L(J,K)
0273      12 U(II,IJ)=A(J,I)+U(II,IJ)
0274      DO 13 I=1,M
0275      DO 13 J=1,M
0276      IL=L(I,I)
0277      13 U(IL,J)=2.*U(IL,J)
0278      RETURN
0279      END

```

\*\* NO ERRORS\*\*      PROGRAM = 00181      COMMON = 00000

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PAGE 0001

FTN4 COMPILER: HP24177 (SEPT. 1974)

```
0280 SUBROUTINE EIVAL(A,N,D,COF,R1,R2)
0281 DIMENSION A(5,5),D(6),COF(6),R1(6),R2(6)
0282 CALL COEF1(A,N,D)
0283 M1=N+1
0284 CALL POLY(D,COF,N,R1,R2,IER)
0285 IF (IER.EQ.0) GO TO 10
0286 WRITE(6,100) IER
0287 100 FORMAT(1X,13#ERRR CODE = ,I2)
0288 10 CONTINUE
0289 RETURN
0290 END
```

\*\* NO ERRORS\*\* PROGRAM = 00064 COMMON = 00030

PAGE 0001

FTN4 COMPILER: HP24177 (SEPT. 1974)

```
0291 SUBROUTINE UNIT1(R,N)
0292 DIMENSION R(5,5)
0293 DO 9 K=1,N
0294 DO 9 J=1,N
0295 9 R(K,J)=0.
0296 DO 8 K=1,N
0297 8 R(K,K)=1.
0298 RETURN
0299 END
```

\*\* NO ERRORS\*\* PROGRAM = 00072 COMMON = 00000

PAGE 0001

FTN4 COMPILER: HP24177 (SEPT. 1974)

```
0300 SUBROUTINE COEF1(A,N,D)
0301 DIMENSION A(5,5),B(5,5),C(5,5),D(6)
0302 CALL UNIT1(B,N)
0303 DO 9 K=1,N
0304 CALL PROD1(A,B,N,N,N)
0305 CALL TRAC1(B,N,5)
0306 L=N-K+1
0307 D(L)=-1./K!*S
0308 CALL UNIT1(C,N)
0309 D1=D(L)
0310 CALL SCH1(C,D1,N)
0311 9 CALL ADJ1(C,B,N)
0312 L=N+1
0313 D(L)=1.
0314 RETURN
0315 END
```

\*\* NO ERRORS\*\* PROGRAM = 00211 COMMON = 00000

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```

0316      SUBROUTINE SEM1(A,D,N)
0317      DIMENSION A(5,5)
0318      DO 9 K=1,N
0319      DO 9 J=1,N
0320      9 A(K,J)=D*A(K,J)
0321      RETURN
0322      END

```

\*\* NO ERRORS\*\* PROGRAM = 00057 COMMON = 00000

PAGE 0001 FTN4 COMPILER: HP24177 (SEPT. 1974)

```

0323      SUBROUTINE TRAC1(A,N,S)
0324      DIMENSION A(5,5)
0325      S=0.
0326      DO 8 K=1,N
0327      8 S=S+A(K,K)
0328      RETURN
0329      END

```

\*\* NO ERRORS\*\* PROGRAM = 00044 COMMON = 00000

PAGE 0001 FTN4 COMPILER: HP24177 (SEPT. 1974)

```

0330      SUBROUTINE ADD1(A,B,N)
0331      DIMENSION A(5,5),B(5,5)
0332      DO 6 K=1,N
0333      DO 6 J=1,N
0334      6 B(K,J)=B(K,J)+A(K,J)
0335      RETURN
0336      END

```

\*\* NO ERRORS\*\* PROGRAM = 00066 COMMON = 00000

PAGE 0001 FTN4 COMPILER: HP24177 (SEPT. 1974)

```

0337      SUBROUTINE PROD1(A,B,N,I,H)
0338      DIMENSION A(5,5),B(5,5),C(5,5)
0339      S=0.
0340      DO 9 K=1,N
0341      DO 9 L=1,H
0342      S=0.
0343      DO 10 J=1,I
0344      10 S=S+A(K,J)*B(J,L)
0345      9 C(K,L)=S
0346      DO 8 K=1,N
0347      DO 8 L=1,H
0348      8 B(K,L)=C(K,L)
0349      RETURN
0350      END

```

\*\* NO ERRORS\*\* PROGRAM = 00183 COMMON = 00000

PAGE 0002 PROD1 FTN4 COMPILER: HP24177 (SEPT. 1974)

0351 END\*

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