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## NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

## Technical Memorandum 33-767

## Transformer Design Tradeoffs

Colonel W. T. McLyman

(NASA-CR-146554) TRANSFORMER DESIGN TRADEOFFS (Jet Propulsion Lab.) 92 p HC \$5.00 CSCL 22B

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## PREFACE

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## CONTENTS

| Introducti                    | on   | I  |  |  |  |  |
|-------------------------------|--|----|--|--|--|--|
| The Design Problem, Generally |  |    |  |  |  |  |
| The Area Product (Ap)         |  |    |  |  |  |  |
| Relations                     | hip of Ap to Transformer Power Handling Capability | 5  |  |  |  |  |
| Output Po                     | ower vs Input Power vs Apparent Power Capability   | 5  |  |  |  |  |
| A Specific                    | c Design Problem as an Example                     | 9  |  |  |  |  |
| Cor                           | e Selection  | 10 |  |  |  |  |
| Win                           | nding Parameters                                   | 14 |  |  |  |  |
| Cur                           | rent Density and Wire Size                         | 15 |  |  |  |  |
| Another I                     | Design Problem as an Example                       | 17 |  |  |  |  |
| Win                           | nding Parameters                                   | 19 |  |  |  |  |
| Cur                           | rent Density and Wire Size                         | 20 |  |  |  |  |
| Reference                     | es   | 23 |  |  |  |  |
| APPEND                        | IXES   |    |  |  |  |  |
| Α,                            | Transformer Power Handling Capability              | 24 |  |  |  |  |
| В.                            | Transformer Efficiency                             | 27 |  |  |  |  |
| C.                            | Relationship of Ap to Control of Temperature Rise  | 29 |  |  |  |  |
| D.                            | Transformer Current Density                        | 36 |  |  |  |  |
| E.                            | Regulation as a Function of Efficiency             | 39 |  |  |  |  |
| F.                            | Window Utilization Factor                          | 4" |  |  |  |  |
| G.                            | Transformer Weight                                 | 49 |  |  |  |  |
| H.                            | Transformer Volume                                 | 52 |  |  |  |  |
| I,                            | Magnetic Core Material Tradeoff                    | 54 |  |  |  |  |
| J.                            | Skin Effect  | 57 |  |  |  |  |
| K.                            | Area Product Ap Relationship                       | 59 |  |  |  |  |

| TABLES  |  |    |
|---------|--|----|
| 1.      | C-Core Characters  | 11 |
| F1.     | Wire Table   | 44 |
| 11.     | Magnetic Core Material Characteristics                       | 55 |
| 12.     | Core Material Characteristics                                | 55 |
| K1.     | Transformer Configuration Constants                          | 59 |
| K2.     | Powder Cores Characteristics                                 | 60 |
| K3.     | Pot Cores Characteristics                                    | 64 |
| K4.     | Laminations Characteristics                                  | 68 |
| K5.     | C-Core Characteristics                                       | 72 |
| K6.     | Tape Wound Core Characteristics                              | 76 |
| FIGURES |  |    |
| 1.      | Transformer Design Factors Flow Chart                        | 3  |
| 2.      | C-Core Transformer   | 4  |
| 3.      | Full Wave Bridge Circuit                                     | 6  |
| 4.      | Full Wave Center Tapped Circuit                              | 7  |
| 5.      | Pushpull Full Wave Center Tapped Circuit                     | 8  |
| 6.      | Magnetic Material Comparison at a Constant Frequency         | 13 |
| B1.     | Transformer Loss Versus Output Load Current                  | 27 |
| Cl.     | Temperature Rise Versus Surface Dissipation                  | 32 |
| C2.     | Surface Area Versus Area Product Ap                          | 33 |
| C3.     | Surface Area Calculation                                     | 35 |
| C4.     | Industrial Description                                       | 35 |
| D1.     | Current Density Versus Surface Area for a 25°C and 50°C Rise | 36 |

## FIGURES

| D2.       | Surface Area Versus Total Watt Loss for a  |            |
|-----------|--|------------|
| <i></i> . | 25°C and 50°C Rise   | 38         |
| E1.       | Transformer Circuit Diagram  | 39         |
| E2.       | Transformer Analytical Equivalent  | 39         |
| F1.       | Resistance Correction Factor (ζ, Zeta) for Wire Temperature Between -50° and 100°C | <b>4</b> 5 |
| G1.       | Transformer Total $W_t$ Versus Area Product $A_p$                                  | 49         |
| Hl.       | C-Core Volume  | 52         |
| H2.       | Transformer Volume Versus Area Product Ap  | 53         |
| II.       | The Typical d.c. B-H Loops of Magnetic Material                                    | 54         |
| Л1.       | Skin Depth Versus Frequency  | 58         |
| J2.       | Skin Depth Equal to AWG Radius Versus Frequency                                    | 58         |
| К1.       | Current Density Versus Area Product Ap for a 25°C and 50°C Rise for Powder Cores   | 62         |
| K2.       | Surface Area Versus Area Product $\mathbf{A}_{\mathbf{p}}$ for Powder Cores        | 62         |
| кз.       | Volume Versus Area Product Ap for Powder Cores                                     | 63         |
| K4.       | Total Weight Versus Area Product Ap for Powder Cores                               | 63         |
| K5.       | Current Density Versus Area Product Ap for a 25°C and 50°C Rise for Pot Cores      | 66         |
| K6.       | Surface Area Versus Area Product Ap for Pot Cores                                  | 66         |
| K7.       | Volume Versus Area Product Ap for Pot Cores  | 67         |
| K8.       | Total Weight Versus Area Product Ap for Pot Cores                                  | 67         |
| К9.       | Current Density Versus Area Product Ap for 25°C and 50°C Rise for Laminations      | 70         |
| k10.      | Surface Area Versus Area Product Ap for Laminations                                | 70         |
| K11.      | Volume Versus Area Product Ap for Laminations                                      | 71         |
| K12.      | Total Weight Versus Area Product Ap for Laminations                                | 71         |
|           |  |            |

## FIGURES

| K13. | Current Density Versus Area Product Ap for 25°C and 50°C Rise for C-Type Cores       | 74 |
|------|--|----|
| K14. | Surface Area Versus Area Product Ap for C-Type Cores                                 | 74 |
| K15. | Volume Versus Area Product Ap for C-Type Cores                                       | 75 |
| K16. | Total Weight Versus Area Product Ap for C-Type Cores                                 | 75 |
| K17. | Current Density Versus Area Product Ap for 25°C and 50°C Rise for Tape-Wound Toroids | 78 |
| K18. | Surface Area Versus Area Product Ap for Tape-Wound Toroids                           | 78 |
| K19. | Volume Versus Area Product Ap for Tape-Wound Toroids                                 | 79 |
| K20. | Total Weight Versus Area Product Ap for Tape-Wound Toroids                           | 79 |
| K21. | Tape Wound Core, Power Cores, and Pot Cores Surface Area At                          | 80 |
| K22. | Lamination Surface Area At   | 80 |
| K23, | C-Core Surface Area At   | 80 |
| K24. | Tape Wound Core, Powder Core, and Pot Cores  | 81 |
| K25. | Laminations  | 81 |
| K26. | C Type Cores   | 81 |

## LIST OF SYMBOLS

| Ac   | effective iron area, cm <sup>2</sup>             |
|--|--|
| A <sub>p</sub><br>A <sub>t</sub><br>A <sub>w</sub> | area product, $W_a \times A_c$ , cm <sup>4</sup> |
| $\mathbf{A}_{t}^{\mathbf{r}}$                      | surface area of a transformer, cm2               |
| $\mathbf{A}_{\mathbf{w}}$                          | wire area, cm <sup>2</sup>                       |
| AWG  | American Wire Gauge                              |
| $^{\mathrm{B}}\mathrm{m}$                          | flux density, teslas                             |
| E  | voltage  |
| η  | efficiency                                       |
| f  | frequency, Hz                                    |
| I  | current, amps                                    |
| ĭo   | load current, amps                               |
|  | primary current, amps                            |
| I<br>P<br>I<br>S<br>J                              | secondary current, amps                          |
|  | current density, amps/cm <sup>2</sup>            |
| $\mathbf{J}_{\mathbf{p}}$                          | primary current density, amps/cm <sup>2</sup>    |
| J <sub>s</sub>                                     | secondary current density, amps/cm2              |
| K  | constant   |
| $^{ m K}{}_{ m j}$                                 | current density coefficient                      |
| ${f K_s}$  | surface area coefficient                         |
| $^{\rm K}{}_{ m u}$                                | window utilization factor                        |
| $^{	extsf{K}}\mathbf{v}$                           | volume coefficient                               |
| $K_{\mathbf{w}}$                                   | weight coefficient                               |
| $\ell_{ m m}$                                      | magnetic path, cm                                |
| l  | linear dimension, cm                             |
| MLT  | mean length turn, cm                             |
| N  | turns  |
| P  | power, watts                                     |
| Pcu  | copper loss, watts                               |
| ${ m P_{fe}}$                                      | core loss, watts                                 |
| $\mathtt{P}_{\mathtt{in}}$                         | input power, watts                               |
| $\mathtt{P}_{o}$                                   | output power, watts                              |
| $\Psi$   | watts/unit area, cm                              |
|  |  |

## LIST OF SYMBOLS (cont)

| P                                      | primary loss, watts                                     |
|--|---|
| $\mathtt{P}_{\mathtt{s}}$              | secondary loss, watts                                   |
| ${\tt P}_{\Sigma}$                     | total loss (core and copper), watts                     |
| P <sub>t</sub>                         | apparent power, watts                                   |
| Ŕ                                      | resistance, ohms  |
| $\mathtt{R}_{\mathbf{E}}$              | equivalent core-loss (shunt) resistance, ohms           |
| Rcu                                    | copper resistance, ohms                                 |
| Reg (%)                                | transformer regulation in percent                       |
| $^{ m R}{}_{ m o}$                     | load resistance, ohms                                   |
| R<br>_ p                               | primary resistance, ohms                                |
| R <sub>s</sub>                         | secondary resistance, ohms                              |
| R <sub>ft</sub>                        | total resistance, ohms                                  |
| $s_1$                                  | conductor area/wire area                                |
| $s_2^-$                                | wound area/usable window                                |
| $s_3^-$                                | usable window area/window area                          |
| $\mathtt{s}_{4}^{\mathtt{s}}$          | usable window area/usable window area + insulation area |
| T                                      | teslas  |
| $v_o$                                  | load voltage, volts                                     |
| Vo1                                    | volume, cm <sup>3</sup>                                 |
| Wa                                     | window area, cm <sup>2</sup>                            |
| $\mathbf{w}_{\mathbf{t}}^{\mathbf{r}}$ | weight, grams   |

#### ABSTRACT

The adoption by NASA of the metric system for dimensioning to replace the long-used English units imposes a requirement on the U.S. transformer designer to convert from the familiar unit to the less familiar metric equivalents. Material is presented to assist in this transition in the field of transformer design and fabrication.

The conversion process in power electronics requires the use of transformer components which frequently are the heaviest and bulkiest items in the conversion circuit. They also have a significant effect upon the overall performance and efficiency of the system. Accordingly, the design of such transformers has an important effect on overall system weight, power-inversion efficiency, and cost.

For years manufacturers have rated their cores with a number that represents its relative power-handling ability. This method assigns to each core a number which is the product of its window area and core cross-section area, and is called "Area Product  $\mathbf{A}_{\mathbf{n}}$ ."

The author has developed a coordination between the  $A_p$  numbers and current density J for a given regulation and temperature rise. The area product  $A_p$  is a dimension to the fourth power, whereas volume is a dimension to the third power and surface area  $A_t$  is a dimension to the second power. The author has developed straight-line relationships for  $A_p$  and Volume,  $A_p$  and surface area  $A_t$  and,  $A_p$  and weight. These relationships can now be used as new tools to simplify and standardize the process of transformer design. They also make it possible to design transformers of small bulk and volume or to optimize efficiency.

#### INTRODUCTION

The conversion process in power electronics requires the use of transformers, components which frequently are the heaviest and bulkiest item in the conversion circuits. They also have a significant effect upon the overall performance and efficiency of the system. Accordingly, the design of such transformers has an important influence on overall system weight, power conversion efficiency and cost. Because of the interdependence and interaction of parameters, judicious design tradeoffs are necessary to achieve optimization.

The information presented herein explains the reasons for making such tradeoffs as a guide for making them intelligently.

Manufacturers have for years assigned numeric codes to their cores which represent the relative power handling ability. This method assigns to each core a number which is the product of its window area and core cross section area and is called "Area Product", A<sub>D</sub>.

Over the last few months, the author became aware of unique relationships between the "Area Product", Ap, characteristic number for transformer cores and several other important parameters which must be considered in transformer design. These numbers were developed by core suppliers to summarize dimensional and electrical properties of C-cores and are listed in their catalogs. Such numbers are available for more than 200 different C-core sizes and configurations.

The author has developed relationships between the  $A_p$  numbers and current density J for a given regulation and temperature rise. The area product  $A_p$  is a dimension to the fourth power  $\ell^4$ , whereas volume is a dimension to the third power  $\ell^3$  and surface area  $A_t$  is a dimension to the second power  $\ell^2$ . Straight-line relationships have been developed for  $A_p$  and volume,  $A_p$  and surface area  $A_t$  and  $A_p$  and weight.

These relationships can now be used as new tools to simplify and standardize the process of transformer design. They make it possible to design transformers of smaller bulk and volume or to optimize efficiency. While developed specifically for aerospace applications, the information has wider utility and can be used for the design of non-aerospace transformers as well.

Because of its significance, area product, A<sub>p</sub>, is treated extensively. Additionally a great deal of information is presented for the convenience of the designer. Much of the material is in graphical or tabular form to assist the designer in making the tradeoffs best suited for his particular application in a minimum amount of time.

One of the basic steps in transformer design is the selection of the proper core material. To aid in the selection of cores a comparison of five common core materials is presented which illustrates their influence on overall transformer efficiency and weight. The designer should also be aware of the cost difference between core materials of the nickel steel families and the silicon steel family. In many instances, the author has found it possible to achieve suitable designs using low cost, silicon steel C-cores when the proper design tradeoffs are made.

#### THE DESIGN PROBLEM, GENERALLY

The designer is faced with a set of constraints which must be observed in the design of any transformer. One of these is the output power, Po, (operating voltage multiplied by maximum current demand) which the secondary winding must be capable of delivering to the load within specified regulation limits. Another relates to minimum efficiency of operation which is dependent upon the maximum power loss which can be allowed in the transformer. Still another defines the maximum permissible temperature rise for the transformer when used in its intended environment having a defined ambient temperature range.

Other constraints relate to volume occupied by the transformer and particularly in aerospace applications, weight, since weight minimization is an important goal in the design of space flight electronics. Lastly, cost effectiveness is often an important consideration.

Depending upon application, certain of these constraints will dominate. Parameters affecting others may then be traded off as necessary to achieve the most desirable design. It is not possible to optimize all parameters in a single design because of the interaction and interdependence of parameters.

For example, if volume and weight are of great significance, reductions in both often can be effected by operating the transformer at a higher frequency but at a penalty in efficiency. When the frequency cannot be raised, reduction in weight and volume may still be possible by selecting a more efficient core material, but at a penalty of increased cost. Judicious tradeoffs thus must be effected to achieve the design goals.

A flow chart showing the interrelation and interaction of the various design factors which must be taken into consideration is shown in Figure 1.

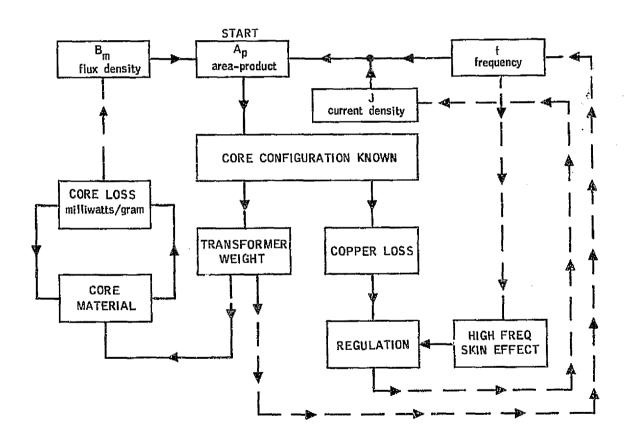


Fig. 1. Transformer Design Factors Flow Chart

Various transformer designers have used different approaches in arriving at suitable designs. For example, in many cases a rule of thumb is used for dealing with current density. Typically, an assumption is made that a good working level is 1000 circular mils per ampere. This may be practical in many

instances but the wire size needed to meet this requirement may produce a heavier and bulkier transformer than desired or required. The information presented herein makes it possible to avoid the use of this and other rules of thumb and to develop a more economical design with great accuracy.

## THE AREA PRODUCT (Ap)

The  $A_p^*$  of a C-type core is the product of the available window area  $(W_a)$  of the core in square centimeters  $(cm^2)$  multiplied by the effective cross-sectional area  $(A_c)$  in square centimeters  $(cm^2)$  which may be stated as:

$$A_{p} = W_{a} A_{c} \qquad \left[cm^{4}\right] \qquad (1)$$

Figure 2 shows in outline form a C-core type transformer typical of those shown in the catalogs of suppliers and uses the letter designations accepted by the industry to indicate certain significant dimensions from which the  $A_p$  area product is calculated. From this it can be seen that  $W_a$  is the FG product and  $A_c$  is the DE product.

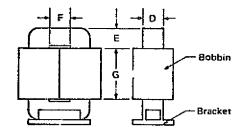
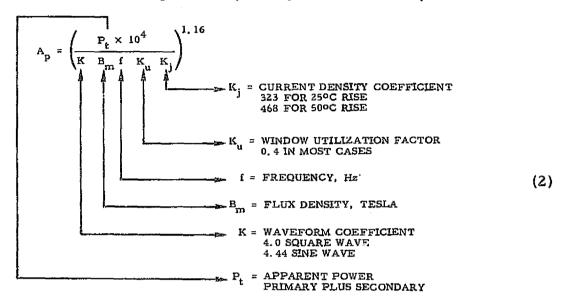


Fig. 2. C-Core Transformer

<sup>\*</sup>Reference 1.

## RELATIONSHIP OF $A_p$ TO TRANSFORMER POWER HANDLING CAPABILITY

According to the newly developed approach, the power handling capability of a core is related to its area product by an equation which may be stated as:



From the above it can be seen that factors such as flux density, frequency of operation, window utilization factor  $K_u$  which defines the maximum space which may be occupied by the copper in the window and the constant  $K_j$  which is related to temperature rise. All have an influence on the transformer area product. The constant  $K_j$  is a new parameter that gives the designer control of the copper loss. Derivation is set forth in detail in Appendix D (page 36).

## OUTPUT POWER VS INPUT POWER VS APPARENT POWER CAPABILITY

Output power  $(P_0)$  is of greatest interest to the user. To the transformer designer it is the apparent power  $(P_t)$  which is associated with the geometry of the transformer that is of greater importance. Assume, for the sake of simplicity, the core of an isolation transformer has but two windings in the window area  $(W_a)$ , a primary and a secondary. Also assume that the window area  $(W_a)$  is divided up in proportion to the power handling capability of the windings using equal current density. The primary winding handles  $P_i$  and the secondary handles  $P_i$  to the load. Since the power transformer has to be designed to accommodate the primary  $P_i$  and secondary  $P_i$ , then:

$$P_{t} = P_{in} + P_{o}$$

$$P_{t} = \frac{P_{o}}{n} + P_{o}$$
(3)

The designer must be concerned with the apparent power handling capability, P<sub>t</sub>, of the transformer core and windings. P<sub>t</sub> may vary by a factor ranging from 2 to 2.828 times the input power, P<sub>in</sub>, depending upon the configuration of the circuit in which the transformer is used because of the different RMS current levels in the windings during operation. If the current wave shape in the rectifier transformer becomes interrupted its effective RMS value changes. Transformer size, thus, is not only affected by the load demand but, also, by the different copper (winding) losses incurred in the various circuit arrangements.

For example, for a load of one watt, compare the power handling capabilities required (neglecting transformer and diode losses so that  $(P_{in} = P_{o})$  for the full-wave bridge circuit of Figure 3, the full-wave center-tapped secondary circuit of Figure 4, and the push-pull center-tapped full-wave circuit in Figure 5.

For the circuit shown in Figure 3,

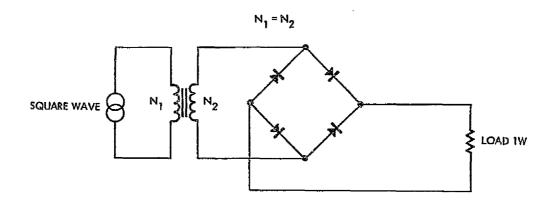


Fig. 3. Full Wave Bridge Circuit

the total apparent power Pt is 2 watts, as may be seen from:

$$P_{t} = (\widehat{I_{N1}} \underbrace{E_{N1}}) + (\widehat{I_{N2}} \underbrace{E_{N2}})$$

$$P_{t} = 2 P_{in}$$
(4)

in which  $I_{N1}$  and  $I_{N2}$  are the currents associated with the primary and secondary windings, respectively, and  $E_{N1}$  and  $E_{N2}$  are the voltages across the primary and secondary windings, respectively.

The circuit shown in Figure 4

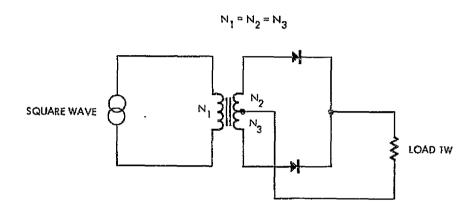


Fig. 4. Full Wave Center Tapped Circuit

requires an increase of 20.7% in  $P_{\rm t}$  due to the increased RMS rating because of the interrupted current flowing in that winding.

$$P_{t} = (I_{N1} E_{N1}) + [(0.707I_{N2} E_{N2}) + (0.707I_{N3} E_{N3})]$$

$$P_{t} = P_{in} + 0.707 P_{in} + 0.707 P_{in} = 2.414 P_{in}$$
(5)

and for the circuit shown in Figure 5

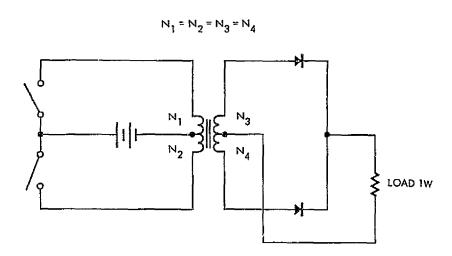


Fig. 5. Pushpull Full Wave Center Tapped Circuit

which is typical of a dc to dc converter, requires a  $P_t$  increase to 2.828 because of the interrupted current flowing in the primary and secondary windings

since 
$$N_1 = N_2 = N_3 = N_4$$
,

$$\mathbf{P_t} = \left[ (0.707 \mathbf{I_{N1}} \ \mathbf{E_{N1}}) + (0.707 \mathbf{I_{N2}} \ \mathbf{E_{N2}}) \right] + \left[ (0.707 \mathbf{I_{N3}} \ \mathbf{E_{N3}}) + (0.707 \mathbf{I_{N4}} \ \mathbf{E_{N4}}) \right] \quad (6)$$

$$P_{t} = 0.707 P_{in} + 0.707 P_{in} + 0.707 P_{in} + 0.707 P_{in} = 2.828 P_{in}$$

Thus the circuit configuration in which the transformer is to be used must be considered by the designer when sizing the transformer.

Rather than discuss the various methods previously used by designers, the author believes it will be more useful to consider typical design problems and to work out solutions using the approach based upon the newly formulated relationships.

## A SPECIFIC DESIGN PROBLEM AS AN EXAMPLE

Assume a specification for a transformer design as shown in Figure 4 (page 7) requiring:

E (output voltage) = 10 volts

I (output current) = 2.0 amps

 $E_{in}$  (input voltage) = 50 volts

\*Operating frequency (f) = 2500 Hz (square wave)

Maximum temperature rise = 25°C

\*\*Transformer efficiency = 95%.

Assuming the bridge rectifier of Figure 3, and using the efficiency constraint of 95%, the apparent power handled by the transformer is calculated (from equation (3)) to be: (1.0 volt diode drop  $(V_d)$  assumed).

Insert values

$$P_{t} = \frac{P_{o}}{\eta} + P_{o}$$
 (3)

$$P_t = I_o (E_o + V_d) \times (\frac{1}{\eta} + 1)$$

$$P_t = \frac{24}{0.95} + 24 = 49.3$$
 watts

This value determines the apparent power handling capability of the core needed for the transformer. A suitable core selection is made by using the area product listings in the catalogs describing the many C-core configurations (sizes and shapes) available from the various suppliers.

<sup>\*</sup>For high frequency skin effect, see Appendix J (page 57).

<sup>\*\*</sup>For transformer regulation as a function of efficiency, see Appendix E (page 39).

## Core Selection

Applying the data from the example to equation (2):

$$A_{p} = \left(\frac{49.3 \times 10^{4}}{(4)(0.3)*(2500)(0.4)(323)}\right)^{1.16} = 1.32 \text{ cm}^{4}$$

After the A<sub>p</sub> has been determined, the geometry of the transformer can be evaluated as described in Appendix G for weight, Appendix C for surface area and Appendix H for volume, and appropriate changes made, if required. Having established the configuration, it is then necessary to determine the core material to complete core selection. Material selection requires consideration of efficiency constraint which is 0.95 in the example. The total transformer losses are

$$P_{\Sigma} = \frac{P_{o}}{n} - P_{o} \tag{7}$$

Inserting values:

$$P_{\Sigma} = \frac{24}{0.95} - 24 = 1.26$$
 watts

Maximum efficiency is realized when the copper (winding) losses are equal to the iron (core) losses (see Appendix B, page 27) which is expressed as

$$P_{cu} = P_{fe}$$
 , and therefore

$$P_{cu} = \frac{P_{\Sigma}}{2}$$
 and thus

$$P_{cu} = 0.63 = P_{fe}$$

Referring to Table 1, column 3 (pages 11 and 12), the AL-124 core with a  $^{A}p$  of 1.44 cm $^{4}$  is closest to the 1.32 cm $^{4}$   $^{A}p$  calculated above.

<sup>&</sup>quot;This is an arbitrary figure developed through years of experience. It can be scaled upwardly for comparison of materials with higher flux density.

# ORIGINAL PAGE TO OF POOR QUALITY

|      | 1            | 2                              | 3                              | 4      | 5                 | 6          | 7      | 8                      | 9   | 10         | 11    | 12           | 13                               | 14           | 15                        | 16                             |
|------|--------------|--------------------------------|--------------------------------|--------|-------------------|------------|--------|------------------------|---|------------|-------|--------------|----------------------------------|--------------|---------------------------|--------------------------------|
|      | Core         | A <sub>t</sub> cm <sup>2</sup> | A <sub>p</sub> cm <sup>4</sup> | MLT cm | N AWG             | l: € 50° C | P<br>2 | $1/\sqrt{\frac{W}{u}}$ | Δ <sup>*</sup> T 25*C<br>J <u>amps</u><br>cm <sup>2</sup> | 11 € 75* C | Py    | ı √ <u>₩</u> | ΔT 50°C<br>J <u>sinps</u><br>cm² | Total Weight | Volume<br>cm <sup>3</sup> | A <sub>c</sub> cm <sup>2</sup> |
|      | AL-2         | 20.4                           | 0, 205                         | 3, 44  | 662 30            | 4, 93      | 0,127  | 0.147                  | 370   | 9. H1      | 1.46  | 0, 273       | 52B                              | 23, 33       | 7, 14                     | 0, 265                         |
| 2    | AL-3         | 23,9                           | 0,410                          | 4.1H   | 662 30            | 10, 5      | 0.717  | 0,18¢.                 | 365   | 11,5       | 1.67  | 0, 269       | 522                              | 31, 18       | B. 92                     | 0, 410                         |
| 3    | AL-÷         | 44,6                           | 0.747                          | 4. 59  | 946 30            | 16, 5      | 1.01   | 0.174                  | 345   | 18.1       | 2.35  | 0.255        | 493                              | 52.8         | 14.06                     | 0, 539                         |
| 4    | ALso         | 37. 4                          | 1.011                          | 5, 23  | 946 30            | 18.8       | 1.13   | 0.1,2                  | 341   | 20.6       | 2.63  | 0. 253       | 489                              | 65.1         | 16.88                     | 0.716                          |
| 5    | AL-124       | 45. 3                          | 1.44                           | 5, 50  | 1317 30           | 27. 5      | 1.36   | 0.157                  | 310   | 30,2       | 3.17  | 0, 229       | 443                              | 80, 8        | 22, 50                    | 0. 716                         |
| 6    | AL-8         | 43.4                           | اد ,2                          | 5, 74  | 221 <sub>20</sub> | 0. 482     | 1.90   | 1,404                  | 271   | 0, 529     | 4,44  | 2, 05        | 395                              | 127, 85      | 35. 66                    | 0, 606                         |
| 7    | AL-4         | 4,9,0                          | 3, 60                          | 4.3K   | 221 20            | 0, 535     | 2,07   | 1. 39                  | 268   | O. 5H7     | 4.83  | 2, 03        | 391                              | 155.8        | 41,62                     | 1, 077                         |
|      | AL-10        | 74.5                           | 3.85                           | 7, 01  | 221 20            | G. 588     | 2,24   | 1.38                   | 266   | 0.646      | 5,22  | 2, 31        | 387                              | 183.2        | 47, 55                    | 1, 342                         |
| "    | AL-12        | N7.0                           | 4, 17                          | 7,49   | 278 20            | 0.748      | 2.61   | 1.32                   | 255   | 0, 821     | 6.09  | . 1, 93      | 371                              | 204, 2       | 61. 38                    | 1, 26                          |
| 10   | AL-139       | 93.7                           | 5, 14                          | 7, 31. | <sup>325</sup> 20 | 0,908      | 2.81   | 1.24                   | 240   | 0.997      | 6.56  | 1.81         | 345                              | 227.0        | 69.63                     | 1, 26                          |
| 111  | AL-7H        | 4H, ]                          | t. c î                         | 7. 01  | 312 20            | 0. R3i     | 2.94   | 1.33                   | 256   | 0,912      | 6.87  | 1.94         | 374                              | 258,0        | 62, 83                    | 1, 34                          |
| 12   | AL-18        | 114                            | 7, 94                          | 7.61   | <sup>310</sup> 20 | 1, 47      | 1. 55  | 1.10                   | 211   | 1,61       | 8.26  | 1.60         | 308                              | 321.0        | P é. 19                   | 1, 45                          |
| 13   | AL-15        | 120                            | 4.07                           | 8.05   | 386 20            | 1.18       | 3.58   | 1.23                   | 237   | 1.30       | 8.40  | 1,79         | 346                              | 352.0        | 94, 43                    | 1,80                           |
| 14   | AL-16        | 127                            | 1v, B                          | 8.89   | 386 20            | 1.30       | 3.80   | 1,20                   | 233   | 1.43       | 8, 89 | 1.76         | 340                              | 397.0        | 104, 95                   | 2, 15                          |
| 15   | AL-17        | 1.42                           | 14, 1                          | 10.3   | 386 20            | 1.51       | 4.25   | 1.185                  | 228   | 1,66       | 9.94  | 1.73         | 333                              | 502.0        | 124, 94                   | 2, 87                          |
| 16   | AL-15        | 159                            | ia,                            | 10.8   | 511 20            | 2,10       | 4.77   | 1.065                  | 205   | 2, 31      | 11.1  | 1.55         | 299                              | 589.0        | 155, 44                   | 2, 87                          |
| 17   | AL-20        | 1821                           | 22,6                           | 11.5   | 511 20            | 2,23       | 5.46   | 1.106                  | 213   | 2.45       | 12.7  | 1.61         | 310                              | 715,0        | 187, 08                   | 3, 58                          |
| tb   | AL-22        | 202                            | 28.0                           | 11,5   | 637 20            | 2.78       | 6.05   | 1.043                  | 201   | 3,05       | 14.1  | 1. 52        | 293                              | 835.0        | 212, 04                   | 3, 58                          |
| -10  | AL-23        | 220                            | 34.9                           | 12.7   | 637 20            | 3.07       | 6.60   | 1.036                  | 200   | 3.37       | 15.4  | 1, 51        | 291                              | 994.0        | 244, 67                   | 4, 48                          |
| 20   | AL-24        | 245                            | 40.D                           | 12,0   | 945 20            | 4. 32      | 7.35   | 0.922                  | 178   | 4,74       | 17.1  | 1, 35        | 259                              | 1090.0       | 280. 91                   | 3, 58                          |
| copp | er loss = it | copper loss : iron loss        |                                |        |                   |            |        |                        |   |            |       |              |                                  |              |                           |                                |

C-Core Characteristics

Table 1.

## Definitions for Table 1

Information given is listed by column as:

- 1. Manufacturer part number
- 2. Surface area calculated from Figure C3
- 3. Area product effective iron area times window area
- 4. Mean length turn on one bobbin
- 5. Total number of turns and wire size for two bobbins using a window utilization factor  $K_{ij} = 0.40$
- 6. Resistance of the wire at 50°C
- 7. Watts loss is based on Figure C1 for a ΔT of 25°C with a room ambient of 25°C surface dissipation times the transformer surface area, total loss is equal to 2 P<sub>C1</sub>
- 8. Current calculated from column 6 and 7
- 9. Current density calculated from column 5 and 8
- 10. Resistance of the wire at 75°C
- 11. Watts loss is based on Figure C1 for a  $\Delta T$  of 50°C with a room ambient of 25°C surface dissipation times the transformer surface area, total loss is equal to 2  $P_{C1}$
- 12. Current calculated from column 10 and 11
- 13. Current density calculated from column 5 and 12
- 14. Effective core weight plus copper weight
- 15. Transformer volume calculated from Figure H1
- 16. Core effective cross-section

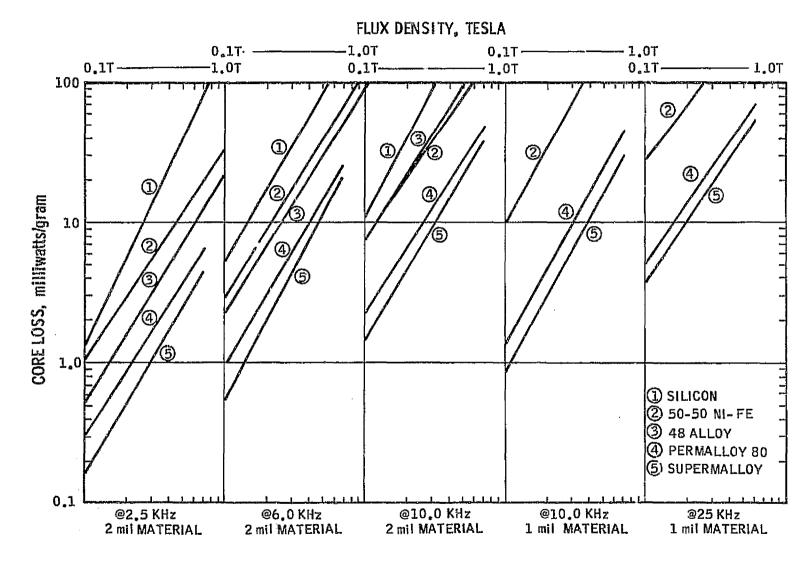


Fig. 6. Magnetic Material Comparison at a Constant Frequency

Referring to column 14, the weight of the core is 46.6 grams. The core loss in milliwatts per gram is obtained from

$$\frac{0.63 \text{ watts}}{46.6 \text{ grams}} = 0.0135 \text{ which converts to}$$

13.5 milliwatts/gram.

The efficiency of various silicon and nickel steels for various high frequencies and flux density is shown in the graphs of Figure 6\*. Reading from the 2.5 KHz frequency curve for a flux density of 0.3 tesla, the loss per gram is about 12 milliwatts per gram, which for 46.6 grams is a total core loss of 560 milliwatts which permits use of a silicon steel core material.

## Winding Parameters

The power loss in the winding can now be accurately determined. First it is necessary to calculate the number of turns in the primary and secondary.

The number of primary turns is calculated from the Faraday law which states:

$$N = \frac{E \times 10^4}{4B_m A_c f}$$
 (8)

Inserting values from the data:

$$N = \frac{50 \times 10^4}{(4)(0.3)(0.716)(2500)} = 233 \text{ turn (primary)}$$

<sup>\*</sup>These curves are for sine waves but are substantially the same for square waves.

(The core cross-section value  $A_c$  is obtained from Table 1 (pages 11 and 12).) The secondary turns are calculated from:

$$\frac{\text{Primary turns}}{\text{Voltage}} = \frac{233}{50} = 4.7 \text{ turns per volt}$$

Since the specified load voltage is 10 volts plus two diode drops,  $4.7 \times 12 = 57$  turns (secondary).

## Current Density and Wire Size

The relationship between the area product  $A_{\mathbf{p}}$  and current density is:

$$J = K_{j} A_{p}^{-0.14}$$
 (9)

in which K is a constant which has a value of 323 for a 25°C rise and a value of 68 for a 50°C rise. Derivation is shown in Appendix D (page 36).

Inserting values:

$$J = (323)(1.44)^{-0.14} = 307 \text{ amp/cm}^2$$

The primary winding current will be:

$$\frac{\text{input power}}{\text{input voltage}} = \frac{25.2}{50.0} = 0.50 \text{ amp}$$

The wire size for the primary is:

$$\frac{0.50}{307} = 0.00162 \text{ cm}^2$$

From the wire table, page 45, No. 25 wire has a diameter of 0.001623 cm<sup>2</sup> and is therefore suitable.

The wire size for the secondary is:

$$\frac{2.0}{307} = .00651 \text{ cm}^2$$

From the wire table, page 45, No. 19 wire has a diameter of 0.00653 cm<sup>2</sup> and is therefore suitable.

The power loss in the windings then can be calculated. The resistance of a winding is the mean length turn in cm multiplied by the resistance in microhms per cm and the total number of turns, or:

$$R = MLT \times N \times (Column C) \times 10^{-6}$$
 [\Omega] (10)

For the primary winding:

$$R = 5.5 \times 0.00106 \times 233 = 1.36 \Omega$$

For the secondary winding:

$$R = 5.5 \times 0.000264 \times 57 = 0.0827\Omega$$

Since power loss is: 
$$P = I^2R$$
 (11)

Copper loss in the primary is  $(0.50)^2 \times 1.36$  or 0.340 watt. In the secondary, the loss is  $(2.0)^2 \times 0.0827$  or 0.331. The total loss in the windings is 0.671 watt. Since the power loss in the core is 0.560 watt, the total power loss in the transformer will be 1.23 watts, which will meet the required efficiency parameter.

## Another Design Problem As An Example

Assume a specification for a transformer design as shown in Figure 4 in which:

E = 56.0 volts after a diode drop 1.0 volt

 $P_0 = 100$  watts to the load

 $E_{IN} = 200 \text{ volts}$ 

\*Operating frequency = 10 KHz (square wave)

Maximum temperature rise = 25°C

\*\*Transformer efficiency = 98%

Because of the diode drop, the actual output power of the transformer is 101.8 watts. Since Figure 4 shows a center tapped secondary, P, is 20.7% greater than in the first example because of the increased RMS rating as explained in equation (5). Thus

$$P_{t} = \left(\frac{P_{o}}{\eta} + P_{o}\right) \times 1.207$$

Inserting values:

$$P_t = \left(\frac{201.8}{0.98} + 101.8\right) \times 1.207 = 248 \text{ watts}$$

The proper core is obtained from the area product using equation (2).

Inserting values:

$$A_{p} = \left(\frac{248 \times 10^{4}}{(4.0)(0.3)(10^{4})(0.4)(323)}\right)^{1.14} = 1.71 \text{ cm}^{4}$$

<sup>\*</sup>For high frequency skin effect, see Appendix J (page 57).
\*\*For transformer regulation as a function of efficiency, see Appendix E (page 39).

After the A<sub>p</sub> has been determined, the geometry of the transformer can be evaluated as described in the first example, (page 10), and appropriate changes made, if desired. Having established the configuration, it is then necessary to determine the core material to complete core selection. Material selection requires consideration of efficiency constraint which is 0.98 in the example.

The transformer losses are, from equation (7)

$$P_{\Sigma} = \frac{P_o}{\eta} - P_o$$

Inserting values:

$$P_{\Sigma} = \frac{101.8}{0.98} - 101.8 = 2.08 \text{ watts}$$

Again maximum efficiency is realized when the copper (winding) losses are equal to the iron (core) losses which is expressed as:

$$P_{cu} = P_{fe}$$
 , and therefore

$$P_{cu} = \frac{P_{\Sigma}}{2}$$
 and thus

$$P_{cu} = 1.04 = P_{fe}$$

Referring to Table 1, column 3 (pages 11 and 12), the AL-8 core with an A<sub>p</sub> of 2.31 is closest to the 1.71 cm<sup>4</sup> A<sub>p</sub> caluclated above. Referring to column 14, the weight of the core is 66.6 grams. The core loss in milliwatts per gram is obtained from

$$\frac{1.04 \text{ watts}}{66.6 \text{ grams}} = 0.0156 \text{ which converts to}$$

Knowing the core loss in milliwatts/grams, the designer refers to the graphs of Figure 6 (page 13). Reading from the curve for the 10 KHz frequency of operation which is specified, it appears that for a flux density of 0.3 tesla, the material that comes closest to 15.6 milliwatts per gram is Permalloy 80 which is approximately 12 milliwatts per gram. When nickel steel is used, Table I2 (page 55) in Appendix I provides a weight correction factor. The weight of 66.6 is increased to 76.5 to give a total core loss of 918 milliwatts.

#### Winding Parameters

The power loss in the winding can then be determined. First it is necessary to calculate the number of turns in the primary and secondary. The number of primary turns is calculated from the Faraday law equation (8) which states:

$$N = \frac{E \times 10^4}{4B_m A_c f}$$

Inserting values from the data:

$$N = \frac{200 \times 10^4}{(4)(0.3)(0.806)(10^4)} = 207 \text{ turns (primary)}$$

(The core cross-section value  $A_c$  is obtained from Table 1, pages 11 and 12).

The secondary turns are calculated from:

$$\frac{\text{primary turns}}{\text{voltage}} = \frac{207}{200} = 1.035 \text{ turns per volt}$$

Since the specified secondary voltage is 57,  $1.035 \times 57 = 59$  turns each side of center tap.

## Current Density and Wire Size

The relationship between the area product  $A_{p}$  and current density from equation (9) is:

$$J = K_j A_p^{-0.14}$$

in which K is a constant which has a value of 323 for a 25°C rise and a value of 468 for a 50°C rise. Derivation is shown in Appendix D (page 36).

Inserting values:

$$J = (323)(2.31)^{-0.14} = 287 \text{ amp/cm}^2$$

the primary winding current will be:

$$\frac{\text{input power}}{\text{input voltage}} = \frac{104}{200} = 0.52 \text{ amp.}$$

The wire size for the primary is:

$$\frac{0.52}{285} = 0.00181 \text{ cm}^2$$

From the wire table, (page 45), No. 25 wire has a diameter of 0.001623 cm<sup>2</sup>. The rule is that when the calculated wire size does not fall close to those listed in the table, the next smallest size should be selected.

The wire size for the secondary is:

$$\frac{\text{output current (0.707)}}{287} = \frac{1.79 \times (0.707)}{287} = 0.0044 \text{ cm}^2$$

From the wire table, No. 21 wire has a diameter of  $0.00411~\mathrm{cm}^2$  and is therefore suitable.

The power loss in the winding then can be calculated. From equation (10), (page 16):

$$R = MLT \times N \times (Column C) \times 10^{-6}$$
 [\Omega]

for the primary winding:

$$R = 5.74 \times 0.001062 \times 207 = 1.26 \Omega$$

for the secondary winding:

$$R = 5.74 \times 0.000419 \times 59 = 0.142 \Omega$$

since power loss is:

$$P = I^2R$$

Copper loss in the primary is  $(0.52)^2 \times 1.26$  or 0.341 watts. In the secondary, the loss is  $(1.79 \times 0.707)^2 \times 0.142 \times 2 = 0.455$  watts. The total loss in the winding is 0.796 watts. Since the power loss in the core is 0.918 watts, the total power loss in the transformer will be 1.714 watts, which will meet the required efficiency parameter.

The author has put in Appendix K the area product A<sub>p</sub> relationships between volume, surface area, current density, and weight for pot core, tape wound cores (toroids), power cores, laminations, and C cores. Much of the material is in graphical or tabular form to assist the designer in making the tradeoffs best suited for his particular application in a minimum amount of time.

#### REFERENCES

- 1. McLyman, C., "Design Parameters of Toroidal and Bobbin Magnetics.
  Technical Memorandum 33-651, Pages 12-15 Jet Propulsion Laboratory,
  Pasadena, Ca.
- Blume, L.F., <u>Transformer Engineering</u>, John Wiley & Sons's, Inc., New York, N.Y. 1938. <u>Pages 272-282</u>
- 3. Terman, F.E., Radio Engineers Handbook, McGraw-Hill Book Co., Inc., New York 1943. Pages 28-37

#### APPENDIX A

#### TRANSFORMER POWER HANDLING CAPABILITY

The power handling capability of a transformer can be related to its A p quantity (which is actually its  $W_aA_c$  product where  $W_a$  is the available core window area in cm<sup>2</sup> and A<sub>c</sub> is the effective cross-sectional area of the core in cm<sup>2</sup>), as follows.

A form of the Faraday law of electromagnetic induction much used by transformer designers states:

$$E = K B_m A_c N f \times 10^{-4}$$
 (A1)

(The constant K is taken at 4 for square wave and at 4.44 for sine wave operation.)

It is convenient to restate this expression as:

$$N A_{c} = \frac{E \times 10^{4}}{4 B_{m} f}$$
 (A2)

for the following manipulation.

By definition the window utilization factor is:

$$K_{u} = \frac{N A_{w}}{W_{a}}$$
 (A3)

and this may be restated as:

$$N = \frac{K_u W_a}{A_{uu}}$$
 (A4)

If both sides of the equation are multiplied by  $A_c$ , then:

$$N A_{c} = \frac{K_{u} W_{a} A_{c}}{A_{w}}$$
 (A5)

From equation (A2):

$$\frac{K_u W_a A_c}{A_w} = \frac{E \times 10^4}{4 B_m f}$$
 (A6)

Solving for  $W_a A_c$ :

$$W_a A_c = \frac{E A_w \times 10^4}{4 B_m f K_u}$$
 (A7)

By definition, current density  $J = amp/cm^2$  which may also be stated:

$$J = \frac{I}{A_{w}} \tag{A8}$$

which may also be stated as:

$$A_{W} = \frac{I}{J} \tag{A9}$$

It will be remembered that transformer efficiency is defined as:

$$\eta = \frac{P_o}{P_{in}}$$
 and  $P_{in} = E I$  (A10)

Rewriting equation (A7) as:

$$EA_{w} = 4B_{mf} K_{u} W_{a} A_{c} 10^{-4} = \frac{EI}{J}$$
 (A11)

and since:

$$\frac{EI}{J} = \frac{P_{in}}{J} = \frac{P_{o}}{J\eta}$$
 (A12)

then:

$$W_a^{A_c}\Big|_{total} = W_a^{A_c}\Big|_{Primary} + W_a^{A_c}\Big|_{Secondary}$$

$$W_{a}^{A}c|_{total} = \frac{P_{o} \times 10^{4}}{J_{4}B_{m}f_{u}} + \frac{P_{o} \times 10^{4}}{4B_{m}f_{u}J} = \frac{P_{o} \times 10^{4}}{4B_{m}f_{u}J} (1/\eta + 1)$$
 (A13)

and since

$$P_{t} = \frac{P_{o}}{\eta} + P_{o} \tag{A14}$$

then

$$W_a A_c = \frac{P_t \times 10^4}{4B_m f K_u J}$$
 (A15)

which may also be stated as applied in Appendix D (page 36) Transformer Current Density as:

$$A_{p} = \frac{P_{t} \times 10^{4}}{4 B_{m} f J K_{u}}$$
 (A16)

#### APPENDIX B

#### TRANSFORMER EFFICIENCY

The efficiency rating of a transformer is a measure of the effectiveness of the design. Efficiency is defined as the ratio of the output power  $P_{o}$  to the input power  $P_{in}$ . The difference between the  $P_{o}$  and the  $P_{in}$  is due to losses. The total power loss in a transformer is made up of fixed losses in the core and quadratic losses in the windings or copper. Thus

$$P_{\Sigma} = P_{fe} + P_{cu}$$
 (B1)

where P fe represents the core loss and P represents the copper loss.

Maximum efficiency is achieved when the fixed loss is equal to the quadratic loss as shown by the equations on page 28. Transformer loss versus output load current is shown in Figure Bl, below.

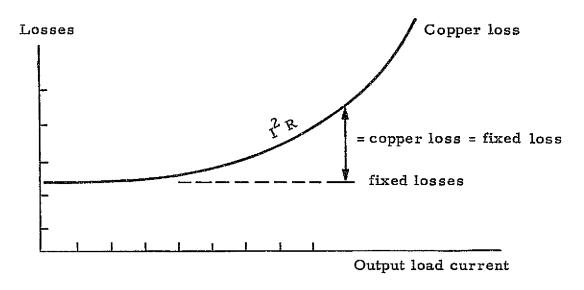


Fig. Bl. Transformer Loss Versus Output Load Current

The copper loss increases as the square of the output power multiplied by a constant K which is thus:

$$P_{cu} = KP_o^2$$
 (B2)

which may be rewritten as

$$P_{\Sigma} = P_{fe} + K P_o^2$$
 (B3)

Since

$$P_{in} = P_{o} + P_{\Sigma}$$

and the efficiency is

$$\eta = \frac{P_o}{P_o + P_{\Sigma}}$$

then:

$$\eta = \frac{P_o}{P_o + P_{fe} + KP_o^2} = \frac{P_o}{P_{fe} + P_o + KP_o^2}$$

and, differentiating with respect to  $P_0$ :

$$\frac{d\eta}{dP_{o}} = -P_{o} \left[ P_{fe} + P_{o} + KP_{o}^{2} \right]^{-2} (I + 2 KP_{o})$$

$$+ \left[ P_{fe} + P_{o} + KP_{o}^{2} \right] = 0 \text{ for max } \eta$$

$$-P_{o} (I + 2 KP_{o}) + \left( P_{fe} + P_{o} + KP_{o}^{2} \right) = 0$$

$$-P_{o} - 2KP_{o}^{2} + P_{fe} + P_{o} + KP_{o}^{2} = 0$$

$$P_{fe} = KP_{o}^{2} = P_{cu}$$
(B4)

#### APPENDIX C

# RELATIONSHIP OF $A_D$ TO CONTROL OF TEMPERATURE RISE

## Temperature Rise

Not all of the P<sub>in</sub> input power to the transformer is delivered to the load as the P<sub>o</sub>. Some of the input power is converted to heat by hysteresis and eddy currents induced in the core material, and by the resistance of the windings. The first is a fixed loss arising from core excitation and is termed "core loss." The second is a variable loss in the windings which is related to the current demand of the load and thus varies as I<sup>2</sup>R. This is termed the quadratic or copper loss.

The generated heat produces a temperature rise which must be controlled to prevent damage to or failure of the windings by breakdown of the wire insulation at elevated temperatures. Such heat is dissipated only from the exposed surfaces of the transformer by a combination of radiation and convection, and thus is dependent upon the total exposed surface area of the core and windings.

Ideally, maximum efficiency is achieved when the fixed and quadratic losses are equal. Thus:

$$P_{\Sigma} = P_{fe} + P_{cu}$$
 (C1)

and

$$P_{CII} = \frac{P_{\Sigma}}{2}$$
 (C2)

When the copper loss in the primary winding is equal to the copper loss in the secondary, the current density in the primary is the same as the current density in the secondary:

$$\frac{P_{p}}{R_{p}} = \frac{P_{s}}{R_{s}}$$
 (C3)

and

$$\frac{P_{\Sigma}}{R_{t}} = \frac{2P_{p}}{R_{p}/2} = \frac{4P_{p}}{R_{p}} = (2I_{p})^{2}$$
 (C4)

then:

$$J_{p} = \frac{I_{p}}{W_{a}/2} = \frac{2I_{p}}{W_{a}} = J_{s} = J$$
 (C5)

# Calculation of Temperature Rise

Temperature rise in a transformer winding cannot be predicted with complete precision, despite the fact that many different techniques are described in the literature for its calculation. One reasonably accurate method for open core and winding construction is based upon the assumption that core and winding losses may be lumped together as:

$$P_{\Sigma} = P_{fe} + P_{cu}$$
 (C6)

and the assumption that thermal energy is dissipated throughout the surface area of the core and winding assembly.

Transfer of heat by <u>radiation</u> occurs because any body raised to a 'emperature above its surroundings emits heat energy in the form of waves. In accordance with the Stefan-Boltzmann law,\* this may be expressed as:

$$W_r = K \epsilon (T2^4 - T1^4) \tag{C7}$$

in which

 $W_{_{\mathtt{T}}}$  = watts per square inch of surface

$$K = 3.68 \times 10^{-11}$$

<sup>\*</sup>Reference No. 2

T2 = hot body temperature in absolute degrees

TI = ambient or surrounding temperature in absolute degrees.

Transfer of heat by <u>convection</u> occurs when a body is hotter than the surrounding medium, which usually is air. A thin layer of air in intimate contact with the hot body is heated by conduction and expands, rising to take the absorbed heat with it. The next layer being colder, replaces the risen layer, and in turn on being heated also rises. This continues until all of the medium surrounding the body is at the body temperature. Transfer of heat by convection\* is stated as:

$$W_{c} = KF\theta^{\eta} \sqrt{p}$$
 (C8)

in which:

W = watts loss per square inch

 $K = 1.4 \times 10^{-3}$ 

F = air friction factor (unity for a vertical surface)

 $\theta$  = temperature rise, degrees C

p = relative barometric pressure (unity at sea level)

 $\eta$  = exponential value ranging from 1.0 to 1.25, depending on the shape and position of the surface being cooled.

The total loss dissipated from a plane vertical surface is expressed by the sum of equations (C7) and (C8),

$$W = 3.68 \times 10^{-11} \epsilon (T2^4 - T1^4) + 1.4 \times 10^{-3} F\theta^{1.25} \sqrt{p}$$
 (C9)

# Temperature Rise Versus Surface Area Dissipation

The temperature rise which may be expected for various levels of power loss is shown in the nomograph of Figure Cl below. It is based on equation (C9)

<sup>\*</sup>Reference No. 2

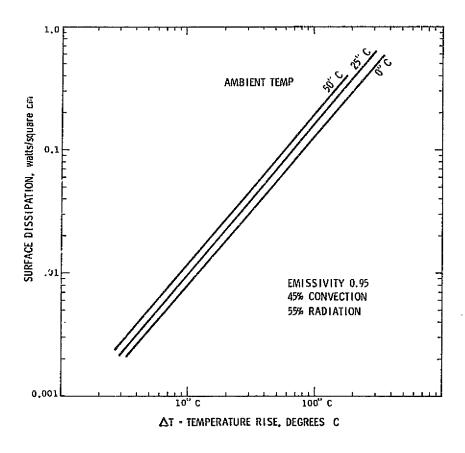


Fig. C1. Temperature Rise Versus Surface Dissipation

relying on data obtained from Reference 2 for heat transfer effected by a combination of 55% radiation and 45% convection, from surfaces having an emissivity of 0.95, in an ambient of 25°C, at sea level. Power loss (heat dissipation) is expressed in watts/cm<sup>2</sup> of total surface area. Heat dissipation by convection from the upper side of a horizontal flat surface is on the order of 15 to 20% more than from vertical surfaces. Heat dissipation from the underside of a horizontal flat surface depends upon surface area and conductivity.

## Surface Area Required for Heat Dissipation

The effective surface area  $\mathbf{A}_t$  required to dissipate heat (expressed as watts loss per unit area) is:

$$A_{t} = \frac{P_{\Sigma}}{\Psi} \tag{C10}$$

in which  $\Psi$  is the power density or the average power lost per unit area of the heat dissipating surface of the transformer and  $P_{\Sigma}$  is the total power lost or dissipated.

Surface area  $A_t$  of a transformer can be related to the area product  $A_p$  of a C-core transformer. The straightline logarithmic relationship shown in Figure C2 below, has been plotted from the data shown in Table 1 (pages 11 and 12).

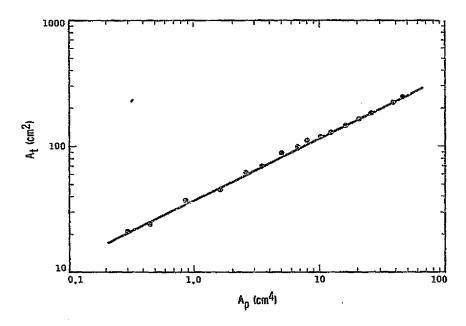
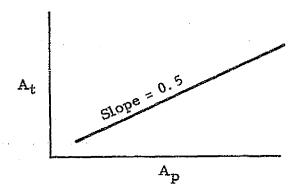


Fig. C2. Surface Area Versus Area Product  $A_p$ 

The relationship is obtained from the conventional slope relationship:

Slope = 
$$\frac{\text{Log A}_{t2}/\text{A}_{t1}}{\text{Log A}_{p2}/\text{A}_{p1}}$$

according to:



in which the subscripts denote the extremes of the values in each column.

From this it appears that:

$$A_{t} = K_{s}(A_{p})^{0.5} = \frac{P_{\Sigma}}{\Psi}$$
 (C11)

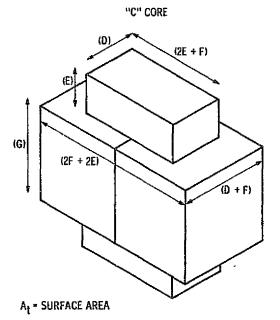
and that (from Fig. C1)

$$\Psi = 0.03 \text{ W/cm}^2 @ 25 ^{\circ}\text{C rise}$$
  
 $\Psi = 0.07 \text{ W/cm}^2 @ 50 ^{\circ}\text{C rise}$ 

in which the constant  $K_s$  has been derived empirically by averaging the data presented in Table 1 (pages 11 and 12) columns 2 and 3. Column 3 was increased to account for the gross area of the iron and  $K_s$  therefore is 39.2.

# Calculation of Surface Area of C-Cores

Table 1 (pages 11 and 12) is a tabulation of data relating to selected C-cores of standard manufacture. The surface areas A<sub>t</sub> of those cores were calculated in accordance with the dimensional relations shown in Figures C3 and C4 below, which derive from the geometry of the core and windings of C-type core transformers as fabricated to industry standards.



 $A_{\xi} = 4E(2E+F) + (ED)(4+2(D+F)(G)+2(2F+2E)(G)+2(D+F)(2F+2E)$ 

Fig. C3. Surface Area Calculation

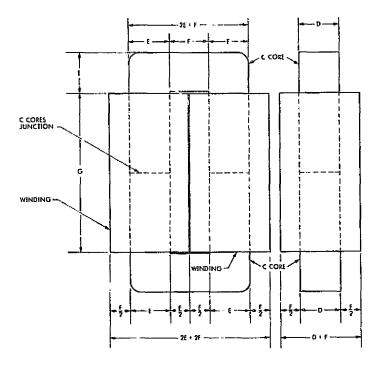


Fig. C4. Industrial Description

#### APPENDIX D

#### TRANSFORMER CURRENT DENSITY

Current density J of a transformer can be related to the surface area A<sub>t</sub> of a C-core transformer for a given temperature rise. The straightline logarithmic relationship shown in Figure Dl below, has been plotted from the data shown in Table 1 (pages 11 and 12).

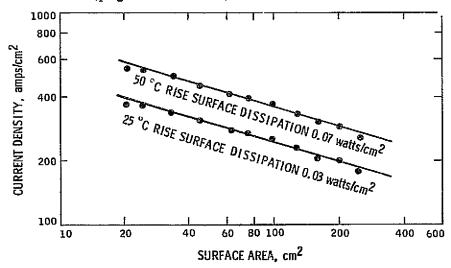
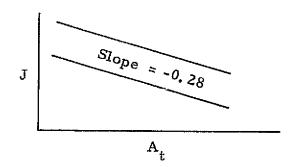


Fig. D1. Current Density Versus Surface Area for a 25°C and 50°C Rise

The relationship is obtained from the conventional slope relationship:

Slope = 
$$\frac{\text{Log } J_1/J_2}{\text{Log } A_t'/A_t^2}$$

according to:



The relationship is:

$$J = K_1 A_t^{-0.28}$$
 (D1)

in which  $K_1$  is a constant which is calculated to be 776 for a 25°C temperature rise and 1120 for a 50° temperature rise.

The relationship of current density J to the area product  $A_{\ p}$  for a given temperature rise can be derived as follows.

The surface area  $\mathbf{A}_t$  relation to the area product  $\mathbf{A}_p$  derived in equation CII of Appendic C, states:

$$A_{t} = K_{s} (A_{p})^{0.5}$$
 (D2)

Combining the equations D1 and D2

$$A_{t}^{-0.28} = \frac{J}{K_{1}} = (K_{s}A_{p}^{0.5})^{-0.28}$$

$$J = K_{1} (K_{s}A_{p}^{0.5})^{-0.28}$$

$$J = K_{1} K_{s}^{-0.28} A_{p}^{-0.14}$$

$$K_{j} = K_{1} (K_{s})^{(-0.28)}$$

$$J = K_{1} A_{p}^{-0.14}$$
(D3)

where:

K<sub>j</sub> for 25°C rise is 323 and K<sub>j</sub> for 50° rise is 468 from the data of Table 1 (pages 11 and 12) in columns 3 and 6 and 3 and 10. This expression may now be inserted in equation (A16) from Appendix A which is:

$$A_{p} = \frac{P_{t} \times 10^{4}}{4B_{m}fJK_{u}}$$

yielding:

1

$$A_{p} = \frac{P_{t} \times 10^{4}}{4B_{m}fK_{u}(K_{j}A_{p}^{-0.14})}$$

$$A_p^{0.86} = \frac{P_t \times 10^4}{4B_m f K_u K_j}$$

$$A_{p} = \left(\frac{P_{t} \times 10^{4}}{4 B_{m} f K_{u} K_{j}}\right)^{1.16} \tag{D4}$$

Figure D2 utilizes the efficiency rating in watts loss in terms of two different, but commonly used allowable temperature rises for the transformer over ambient temperature. The data presented are used as bases for indicating the needed transformer surface area  $A_t$  (in cm<sup>2</sup>).

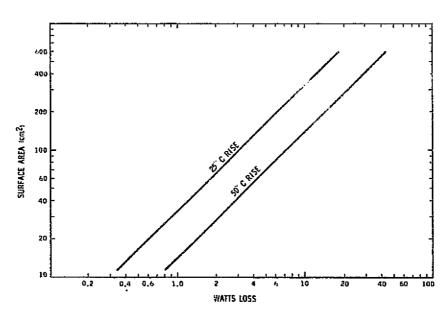


Fig. D2. Surface Area Versus Total Watt Loss for a 25°C and 50°C Rise

#### APPENDIX E

# REGULATION AS A FUNCTION OF EFFICIENCY

The size of a transformer usually is determined either by a temperature rise limit, or by allowable voltage regulation, assuming that size and weight are to be minimized.

Figure El below shows circuit diagram of a transformer with one secondary.

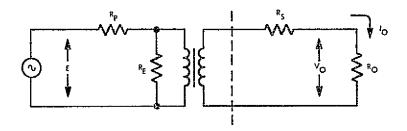


Fig. El. Transformer Circuit Diagram

The analytical equivalent is shown in Figure E2.

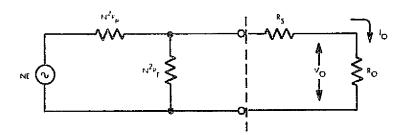


Fig. E2. Transformer Analytical Equivalent

This assumes that distributed capacitance in the secondary can be neglected because the secondary voltage is not excessive. Also the winding

geometry is designed to limit the leakage inductance to a level low enough to be neglected under most operating conditions.

Transformer voltage regulation can be expressed as:

Reg (%) = 
$$\frac{V_O(N_*L_*) - V_O(F_*L_*)}{V_O(N_*L_*)} \times 100$$
 (E1)

in which  $V_{o}$  (N. L.) is the no load voltage and  $V_{o}$  (F. L.) is the full load voltage.

The output voltage computed using Figure El is:

$$V_{o} = \frac{R_{o}}{R_{o} + R_{s}} \frac{(N^{2}R_{p}) \| (N^{2}R_{E}) \| (R_{o} + R_{s})}{N^{2}R_{p}} NE$$
 (E2)

For the usual condition of

$$N^2 R_E \gg N^2 R_p \| (R_o + R_s),$$

V simplifies to

$$V_o = V_o(F.L.) = \frac{R_o}{R_o + (N^2 R_p + R_s)} NE$$
 (E3)

For equal window areas allocated for the primary and secondary windings, it can be shown that  $N^2R_p = R_s$ .

For simplicity

$$R_{cu} = N^2 R_p + R_s = 2R_s$$

At no load (N. L.) R approaches infinity, therefore:

$$V_{O}(N. L.) = NE$$
 (E4)

$$Reg (\%) = \frac{R_o}{R_o + R_{cu}} NE$$

$$Reg (\%) = \frac{R_o}{NE} \times 100$$
(E5)

$$= \left(1 - \frac{R_o}{R_o + R_{cu}}\right) \times 100$$
 (E6)

$$= \frac{R_{cu}}{R_o + R_{cu}} \times 100 \tag{E7}$$

Thus it appears that regulation is independent of the transformer turns ratio.

Regulation as a function of copper loss, multiply the equation E7 by  $I_0^2$ 

Reg (%) = 
$$\frac{I_o^2 R_{cu}}{I_o^2 (R_o + R_{cu})} \times 100$$
 (E8)

then

Reg (%) = 
$$\frac{P_{cu}}{P_0 + P_{cu}} \times 100$$
 (E9)

$$P_{in} = P_{cu} + P_{fe} + P_{o}$$
 (E10)

Regulation as a function of efficiency

$$\frac{P_o}{P_{in}} = \frac{P_o}{P_{cu} + P_{fe} + P_o} = \eta \tag{E11}$$

By definition

$$P_{cu} = P_{fe}$$

Solving for P<sub>cu</sub> + P<sub>fe</sub>

$$\frac{P_{o}(1-\eta)}{\eta} = P_{o}(\frac{1}{\eta}-1) = P_{cu} + P_{fe} = 2 P_{cu}$$
 (E12)

$$\frac{\text{Reg (\%)}}{100} = \frac{1}{1 + \frac{0}{P_{cu}}} = \frac{1}{1 + \frac{2}{1/\eta - 1}} = \frac{1 - \eta}{1 + \eta}$$
 (E13)

Reg (%) = 
$$\frac{1-\eta}{1+\eta} \times 100$$
 (E14)

Efficiency as a function of regulation, multiply both sides of the equation by  $(1 + \eta)$ :

$$Reg (\%) + \eta Reg (\%) = 100 - \eta 100$$
 (E15)

solve for n

$$\eta \ 100 + \eta \ \text{Reg} \ (\%) = 100 - \ \text{Reg} \ (\%)$$
 (E16)

$$\eta (100 + \text{Reg } (\%) = 100 - \text{Reg } (\%)$$
 (E17)

$$\eta = \frac{100 - \text{Reg (\%)}}{100 + \text{Reg (\%)}} \tag{E18}$$

#### APPENDIX F

#### WINDOW UTILIZATION FACTOR

The fraction  $K_u$  of the available core window space which will be occupied by the winding (copper) is calculated from areas  $S_1$ ,  $S_2$ ,  $S_3$ , and  $S_4$ :

$$K_{u} = S_{1} \times S_{2} \times S_{3} \times S_{4}$$
 (F1)

where

$$S_2 = \frac{\text{wound area}}{\text{usable window area}}$$
 and,

$$S_3 = \frac{\text{usable window area}}{\text{window area}}$$

in which

conductor area = copper area

wire area = copper area + insulation area

wound area = number of turns x wire area of one turn

usable window are = available window area minus residual area which
results from the particular winding technique used

window area = available window area

insulation area = area usable for winding insulation

 $S_1$  is dependent upon wire size. Columns A and D of Table F1, page 44 may be used for calculating some typical values such as for AWG 10, AWG 20, AWG 30 and AWG 40.

Table F1. Wire Table

| Awg<br>Wire<br>Size | Bare                 | Ares                 | Resistance | Resistance Peavy Synthetics      |                      |         |                   |        |                   |          |                   |          |
|---------------------|----------------------|----------------------|------------|----------------------------------|----------------------|---------|-------------------|--------|-------------------|----------|-------------------|----------|
|                     | em <sup>2</sup> 10-3 |                      | 10-6 Ω     | A:                               | ea.                  | Diam    | eter              | Turns  | Per               | Turs     | ış-Per            | Weight   |
|                     | (destante b)         | CIR-MIL <sup>®</sup> | cm at 20°C | cm <sup>2</sup> 10 <sup>-3</sup> | CIR-MIL <sup>8</sup> | em.     | inch <sup>a</sup> | cm     | inch <sup>a</sup> | cm²      | iach <sup>2</sup> | gın/cm   |
| 10                  | 52.61                | 10384                | 32.70      | 55, 9                            | 11046                | 0, 267  | 9, 1051           | 3.87   | 9.5               | 10,73    | 69, 20            | 0,468    |
| 11                  | 41.68                | 8226                 | 41.37      | 44,5                             | 8798                 | 0, 238  | 0.0938            | 4, 36  | 10.7              | 13.48    | 89.95             | 0.3750   |
| 12                  | 33.08                | 6529                 | 52,09      | 35.64                            | 7022                 | 0. 213  | 0.0838            | 4. RS  | 11.4              | 16.81    | 108.4             | 0.2977   |
| 13                  | 26.26                | 5184                 | 65.64      | 28.36                            | 5610                 | 0, 198  | 0.0749            | 5.47   | 13.4              | 21.15    | 136.4             | 0,2367   |
| 14                  | 20.82                | 4109                 | 82,80      | 22.95                            | 4556                 | 0. 171  | 0.0675            | 6.04   | 14.8              | 26, 14   | 168.6             | 0,1879   |
| 15                  | 16.51                | 3260                 | 104.3      | 18.37                            | 3624                 | 0, 153  | 0, 0602           | 6,77   | 16.6              | 32.66    | 210,6             | 0, 1492  |
| 16                  | 13.07                | 2581                 | 131.6      | 14.73                            | 2905                 | 0.137   | 0.0539            | 7, 3Z  | 18.6              | 40, 73   | 262.7             | 0.1184   |
| 17                  | 10.39                | 2052                 | 165.8      | 13.68                            | 2323                 | 0, 122  | 0,0482            | 8.18   | 8.05              | 51,36    | 331.2             | 0,0943   |
| 18                  | 8.228                | 1624                 | 209,5      | 9.326                            | 1857                 | 0.109   | 0, 0431           | 9.13   | 23.2              | 64,33    | 414.9             | 0.0747   |
| 19                  | 6.531                | 1289                 | 263.9      | 7.539                            | 1490                 | 0, 0980 | 0, 0386           | 10.19  | 25.9              | 79.85    | 515.0             | 0,05940  |
| 20                  | 5.188                | 1024                 | 332.3      | 6.065                            | 1197                 | 0.0879  | 0.0346            | 11, 37 | 28.9              | 98,93    | 638.1             | 0.04726  |
| 21                  | 4.116                | 812.3                | 416.9      | 4,837                            | 954.8                | 0.0785  | 0.0309            | 12.75  | 32,4              | 124.0    | 799.8             | 0, 0375  |
| 22                  | 3.243                | 640.1                | 531.4      | 3.857                            | 761-7                | 0.0701  | 0.0276            | 14, 25 | 36.2              | 155.5    | 1003              | 0. 0296  |
| 23                  | 2,588                | 510.8                | 666.0      | 3, 135                           | 620, 0               | 0,0632  | 0.0249            | 15.82  | 40, 2             | 191.3    | 1234              | 0.0237   |
| 24                  | 2.047                | 404,0                | 842,1      | 2.514                            | 497.3                | 0.0566  | 0, 0223           | 17.63  | 44.8              | 23B.6    | 1539              | 0.0188   |
| 25                  | 1.623                | 320,4                | 1062.0     | 2.002                            | 396.0                | 0,0505  | 0.0199            | 19.80  | 50.3              | 299.7    | 1933              | 0,0149   |
| 26                  | 1.280                | 252,8                | 1345.0     | 1,603                            | 316.8                | 0.0452  | 0,0178            | 22, 12 | 56.2              | 374,2    | 2414              | 0, 0118  |
| 27                  | 1.021                | 201.6                | 1687.6     | 1.313                            | 259, 2               | 0.0409  | 0.0161            | 24.44  | 62.1              | 456.9    | 2947              | 0.0094   |
| 28                  | 0,8046               | 158.8                | 2142.7     | 1,0515                           | 207.3                | 0.0366  | 0.0144            | 27.32  | 69.4              | 570.6    | 3680              | 0.0074   |
| 29                  | 0.6470               | 127,7                | 2664.3     | 0, 8548                          | 169.0                | 0,0330  | 0.0130            | 30, 27 | 76,9              | 701.9    | 4527              | 0.0060   |
| 30                  | -0.5067              | 100.0                | 3402.2     | 0.6785                           | 134.5                | 0.0294  | 0.0116            | 33.93  | 85.2              | 884,3    | 5703              | 0,0047   |
| 31                  | 0,4013               | 79.21                | 4294.6     | 0.5596                           | 110.2                | 0.0267  | 0,0105            | 37.48  | 95.2              | 1072     | 6914              | 0,0037   |
| 32                  | 0,3242               | 64.00                | 5314.9     | 0.4559                           | 90.25                | 0.0241  | 0,0095            | 41.45  | 105.3             | 1316     | 8488              | 0. 90309 |
| 33                  | 0,2554               | 50.41                | 6748.6     | 0.3662                           | 72, 25               | 0,0216  | 0,0085            | 46.33  | 117.7             | 1638     | 10565             | 0.0024   |
| 34                  | 0.2011               | 39.69                | 8572,8     | 0, 2863                          | 56, 25               | 0.0191  | 0.0075            | 52, 48 | 133.3             | 2095     | 13512             | 0,0018   |
| 35                  | 0.1589               | 31.36                | 10849      | 0.2268                           | 44. 89               | 0.0170  | n. 0067           | 58.77  | 149. 3            | 2645     | 17060             | 0,0015   |
| 36                  | 0,1266               | 25.00                | 13608      | 0. 1613                          | 36,00                | 0.0152  | 0.0060            | 65,62  | 166,7             | 3309     | 21343             | 0.0011   |
| 37                  | 0.1026               | 20.25                | 16801      | 0, 1538                          | 30, 25               | 0,0140  | 0, 0055           | 71.57  | 181.8             | 3901     | 25161             | 0.0009   |
| 38                  | 0.08107              | 16.00                | 21266      | 0.1207                           | 24.01                | 0.0124  | 0.0049            | 80, 35 | 204.1             | 4971     | 32062             | 0.0007   |
| 37                  | 0, 06207             | 12.25                | 27775      | 0.0932                           | 18.49                | 0,0109  | 0.0043            | 91.57  | 232.6             | 6437     | 41518             | 0.0005   |
| 40                  | 0.04869              | 9.61                 | 35400      | 0.0723                           | 14, 44               | 0.0096  | 0.0038            | 103.6  | 263.2             | 8298     | 53522             | 0.0004   |
| 41 /                | 0,03972              | 7.84                 | 43405      | 0.0584                           | 11.56                | 0,00863 | 0.0034            | 115.7  | 294.1             | 10273    | 66260             | 0.0003   |
| 42                  | 0.03166              | 6.25                 | 54429      | 0.04558                          | 9.00                 | 0.00762 | 0.0030            | 131.2  | 333,3             | 13163    | 84901             | 0.0002   |
| 43                  | 0,02452              | 4. 84                | 70308      | 0.03683                          | 7. 29                | 0.00685 | 0.0027            | 145.8  | 370.4             | 16291    | 105076            | 0.0002   |
| 44                  | 0,0202               | 4.00                 | 85072      | 0.03165                          | 6, 25                | 0,00635 | 0,0025            | 157.4  | 400.0             | 18957    | 122272            | 0.0001   |
|                     | A                    | <u></u>              |            | Δ                                | E                    |         | <u></u>           | H      | i                 | <u> </u> | <u> </u>          | <u> </u> |

<sup>&</sup>quot;This data from REA Magnetic Wire Detaintor (Ref. 1).



<sup>&</sup>lt;sup>b</sup>This solution means the entry in the column must be multiplied by 10<sup>-3</sup>

## TEMPERATURE CORRECTION FACTORS

图

The values shown in Fig. I are based upon a correction factor of 1.0 at 20°C. For other temperatures the effect upon wire resistance can be calculated by multiplying the resistance value for the wire size shown in column C of Table 2 by the appropriate correction factor shown on the graph. Thus, Corrected Resistance =  $\mu\Omega/cm$  (at 20°C) x  $\chi$ 

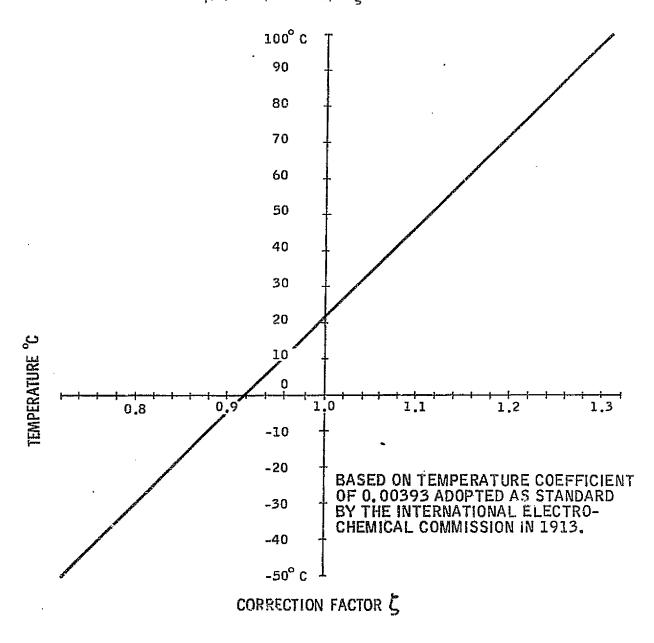


Fig. Fl. Resistance Correction Factor (ζ, Zeta) for wire temperature between -50° and 100°C

## CONVERSION DATA FOR WIRE SIZES FROM #10 to #44

Columns A and B in Table F1 give the bare area in the commonly used circular mils notation and in the metric equivalent for each wire size. Column C gives the equivalent resistance in microhms/centimeter ( $\mu\Omega/\text{cm}$  or  $10^{-6}\Omega/\text{cm}$ .). Columns D to L relate to coated wires showing the effect of insulation on size and the number of turns and the total weight in grams/centimeter.

The total resistance for a given winding may be calculated by multiplying the MLT (mean length/turn) of the winding in centimeters, by the microhms cm for the appropriate wire size (Column C), and the total number of turns. Thus

$$R = (MLT) \times (N) \times (Column C) \times 10^{-6}$$
 [ohms]

The weight of the copper in a given winding may be calculated by multiplying the MLT by the grams/cm (Column L) and by the total number of turns. Thus

$$W_{t} = (MLT) \times (N) \times (Column L)$$
 [grams]

Turns per square inch and turns per square cm are based on 60% wire fill factor.

Thus:

AWG 10 = 
$$\frac{52.61 \text{ cm}}{55.90 \text{ cm}} = 0.941$$
;  
AWG 20 =  $\frac{5.188 \text{ cm}}{6.065 \text{ cm}} = 0.855$ ;  
AWG 30 =  $\frac{0.5067 \text{ cm}}{0.6785 \text{ cm}} = 0.747$ ; and  
AWG 40 =  $\frac{0.04869 \text{ cm}}{0.0723 \text{ cm}} = 0.673$ .

S<sub>2</sub> is the fill factor for the usable window area. It can be shown that for circular cross-section wire wound on a flat form the ratio of wire cm<sup>2</sup> to the area required for the turns can never be greater than 0.91. In practice, the actual maximum value is dependent upon the tightness of winding, variations in insulation thickness, and wire lay. Consequently, the fill factor is always less than the theoretical maximum.

As a typical working value for copper wire with a heavy synthetic film insulation, a ratio of 0.60 may be safely used.

The term  $S_3$  defines how much of the available window space may actually be used for the winding. The winding area available to the designer depends on the bobbin configuration. A single bobbin design offers an effective  $W_a$  between 0.835 to 0.929 while a two bobbin configuration offers an effective  $W_a$  between 0.687 to 0.872. A good value to use for both configurations is 0.75.

The term  $S_4$  can vary from 1.0 to 0.80 and defines how much of the usable window space is actually being used for insulation. If the transformer has multiple secondaries having significant amounts of insulation  $S_4$  could be as low as 0.8.

A typical value for the copper fraction in the window area is about 0.40. For example, for AWG 20 wire,  $S_1 \times S_2 \times S_3 \times S_4 = 0.855 \times 0.060 \times 0.75 \times 1.0 = 0.385$ , which is very close to 0.4.

This may be stated somewhat differently as:

$$0.4 = \frac{A_{w} \text{ Bare}}{A_{w} \text{ Total}} \times \text{Fill Factor} \times \frac{W_{a(eff)}}{W} \times \text{Insulation Factor}$$

$$(S_{1}) \qquad (S_{2}) \qquad (S_{3}) \qquad (S_{4})$$

#### APPENDIX G

## TRANSFORMER WEIGHT

The total weight  $W_t$  of a transformer can be related to the area product  $A_p$ . The straightline logarithmic relationship shown in Figure Gl below, has been calculated from the data shown in Table 1 (pages 11 and 12).

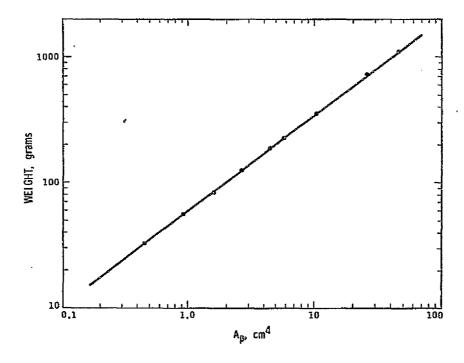


Fig. G1. Transformer Total  $W_t$  Versus Area Product  $A_p$ 

This relationship is obtained from the conventional slope relationship:

Slope = 
$$\frac{\text{Log}(W_{t2}/W_{t1})}{\text{Log}(A_{p2}/A_{p1})}$$

in which the W  $_{\rm t}$  and A  $_{\rm p}$  values are the extremes of the data shown in columns 14 and 15 for weight, and column 3 for area product.

The relationship is:

$$W_t = K_w A_p^{0.75}$$
 (G1)

in which the constant  $K_{w}$  has been derived empirically by averaging the data presented in columns 3, 14 and 15 of Table 1 (pages 11 and 12) and is 66.6.

Table I2 (page 55) shows how weight varies as a function of selected different magnetic materials used for transformer C-cores. Magnetic materials for C-cores are discussed in Appendix I (page 54).

Derivation of the relationship is according to the following: Weight  $W_t$  varies in accordance with the cube of any linear dimension  $\ell$  (designated  $\ell^3$  below), whereas, area product  $A_p$  varies as the fourth power:

$$W_{t} = K_{1} \ell^{3} \tag{G2}$$

$$A_{p} = K_{2} \ell^{4} \tag{G3}$$

$$\ell^4 = \frac{A_p}{K_2} \tag{G4}$$

$$\ell = \left(\frac{A_p}{K_2}\right)^{0.25} \tag{G5}$$

$$\ell^3 = \left[ \left( \frac{A_p}{K_2} \right)^{0.25} \right]^3 = \left( \frac{A_p}{K_2} \right)^{0.75}$$
 (G6)

$$W_{t} = K_{1} \left(\frac{A_{p}}{K_{2}}\right)^{0.75} \tag{G7}$$

$$K_{W} = \frac{K_{1}}{K_{2}^{0.75}} \tag{G8}$$

$$W_{t} = K_{w} A_{p}^{0.75}$$
 (G9)

in which  $\mathbf{K}_1$  is a constant depending upon the core material, and  $\mathbf{K}_2$  is related to core and window dimensions.

#### APPENDIX H

#### TRANSFORMER VOLUME

The volume of a transformer can be related to the area product A of a C-core transformer, treating the volume as shown in Figure H1 below as a solid cube quantity without subtraction of anything for the core window.

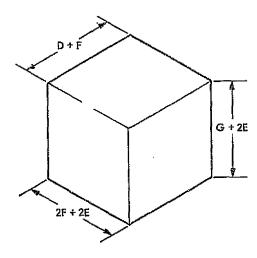


Fig. Hl. C-Core Volume

The straight-line logarithmic relationship plotted in Figure H2 below, has been calculated from data in Table 1, using the data shown in Figure H1 above.

The relationship is obtained from the conventional slope relationship:

Slope = 
$$\frac{\text{Log}(\text{Vol. 2/Vol. 1})}{\text{Log}(A_p^2/A_p^1)}$$

in which the Vol. and  $A_p$  values are the extremes of the data shown in column 15 for volume, and column 3 for area product.

The volume/area product relationship is:

Vol. = 
$$K_v A_p^{0.75}$$
 (H1)

in which  $K_v$  is a constant related to core configuration. It is 17.9 for a C-core, which has been derived by averaging the values in Table 1.

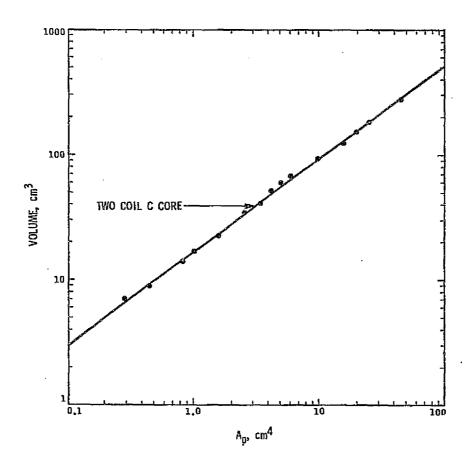


Fig. H2. Transformer Volume Versus Area Product  $A_p$ 

#### APPENDIX I

# MAGNETIC CORE MATERIAL TRADEOFF

The relationships between area product  $A_p$  and certain parameters are associated only with such geometric properties as surface area and volume, weight, and the factors affecting temperature rise such as current density.  $A_p$  has no relevance to the magnetic core materials used, but since the designer often must make tradeoffs between such goals as efficiency and size which are influenced by core material selection, some useful data is presented below.

In the many articles written about inverter and converter transformer design, recommendations with respect to choice of core material usually are a compromise selection of material characteristics such as those tabulated in Table II, and graphically displayed in Figure II. The selected data are typical of commercially available core materials suitable for the mentioned applications.

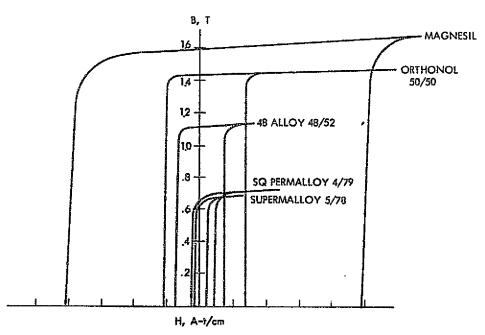


Fig. II. The Typical d.c. B-H Loops of Magnetic Material

Table II. Magnetic core material characteristics

| Trade names                                     | Composition               | Saturated<br>flux<br>density,<br>T <sup>1</sup> | DC coercive<br>force,<br>amp-turn/<br>cm | Squareness<br>ratio | Material<br>density,<br>g/cm <sup>3</sup> | Loss factor<br>at 3 kHz and<br>0.5 T, W/kg |
|---|---------------------------|---|--|---------------------|---|--|
| Magnesil<br>Silectron<br>Microsil<br>Supersil   | 3% Si<br>97% Fe           | 1,5-1.8   | 0, 5-0, 75                               | 0.85-1.0            | 7.63                                      | 33, 1                                      |
| Deltamax<br>Orthonol<br>49 Sq. Mu               | 50% Ni<br>50% Fe          | 1.4-1.6   | 0.125-0.,25                              | 0,94-1.0            | 8. 24                                     | 17.66                                      |
| Allegheny 4750<br>48 Alloy<br>Carpenter 49      | 48% Ni<br>52% Fe          | 1, 15-1, 4                                      | 0.062-0.187                              | 0.80-0.92           | 8.19                                      | 11.03                                      |
| 4-79 Permailoy<br>Sq. Permalloy<br>80 Sq. Mu 79 | 79% Ni<br>17% Fe<br>4% Mo | 0.66-0.82                                       | 0.025-0.05                               | 0.80-1.0            | 8,73                                      | 5, 51                                      |
| Supermalloy                                     | 78% Ni<br>17% Fe<br>5% Mo | 0.65-0.82                                       | 0.0037-0.01                              | 0.40-0.70           | 8.76                                      | 3. 75                                      |

 $<sup>^{2}</sup>$ 1 g/cm<sup>3</sup> = 0.036 lb/in.<sup>3</sup>

Table I2. Core material characteristics

| Material     | Density | Factor* |
|--------------|---------|---------|
| Magnesil     | 7, 63   | 1,000   |
| Supermender  | 8. 15   | 1,066   |
| 48 Alloy     | 8.19    | 1.073   |
| Orthonol     | 8, 24   | 1.079   |
| Sq Permalloy | 8.73    | 1.144   |
| Supermalloy  | 8.77    | 1.148   |

<sup>\*</sup>Weight factor.

As can be seen, the material which provides the highest flux density, silicon, produces the smallest component size. If size is the most important consideration, this would determine the choice of materials. On the other hand, the type 78 SUPERMALLOY material (see the 5/78 curve in Figure II), has the lowest flux density and this material would result in the largest size transformer. However, this material has the lowest coercive force and lowest core loss of any of the available materials. These factors might well be decisive in other applications.

Inverter transformer design usually is aimed at achieving the smallest size with the highest efficiency, and with adequate performance for the widest range of environmental conditions. Unfortunately, the material which produces the smallest size has the lowest efficiency, and conversely, the highest efficiency materials result in the largest size. Thus tradeoffs must be made between the allowable transformer size and the minimum tolerable efficiency. Choice of core material is thus based upon achieving the best characteristic for the most critical or important design parameter, with acceptable compromises on all other parameters.

Fortunately, there is such a wide choice of core sizes available (Table 1, pages 11 and 12, lists only 20 out of more than 200 commercially available), that relative proportions of iron and copper can be varied without changing the  $A_{\rm p}$  area product.

<sup>\*</sup>However, at frequencies above about 20 kHz, eddy current losses are so much greater than hysteresis losses that it is necessary to use very thin (1 and 2 mil) strip cores.

#### APPENDIX J

#### SKIN EFFECT

## Skin Effect

It is now common practice to operate dc-to-dc converters at frequencies up to 50 kHz. At higher frequencies, skin effect alters the predicted efficiency since the current carried by a conductor is distributed uniformly across the conductor cross-section only at dc and at low frequencies. The concentration of current near the wire surface at higher frequencies is termed the skin effect. This is the result of magnetic flux lines which circle only part of the conductor. Those portions of the cross section which are circled by the largest number of flux lines exhibit greater reactance.

Skin effect accounts for the fact that the effective alternating current resistance to direct current ratio is greater than unity\*. The magnitudes of the effects due to increased frequency on conductivity, magnetic permeability and inductange are sufficient to require further consideration of the size of the conductor. The depth of the skin effect is expressed by:

depth (cm) = 
$$6.61/f^{1/2}$$
) K (J1)

in which K is a constant according to the relationship:

$$K = [(1/\mu r) \rho/\rho c]^{1/2}$$
 (J2)

in which:

 $\mu r$  = relative permeability of conductor material ( $\mu r$  = 1 for copper and other nonmagnetic materials)

ρ = resistivity of conductor material at any temperature

ρc = resistivity of copper at 20°C = 1.724 microhm-centimeter

K = unity for copper

<sup>\*</sup>Reference 3.

Figures J1 and J2 below show respectively, skin depth as a function of frequency according to equation (J2) above, and as related to the AWG radius, or as  $R_{ac}/R_{dc}$  = 1 versus frequency.\*

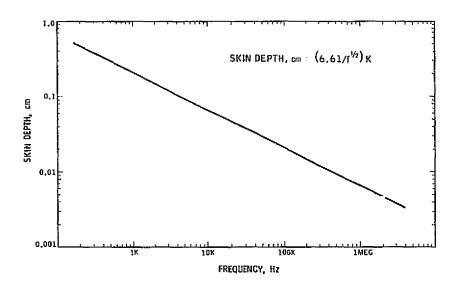


Fig. Jl. Skin Depth Versus Frequency

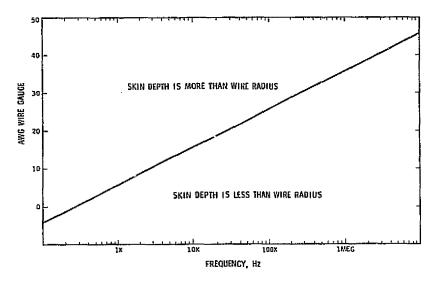


Fig. J2. Skin Depth Equal to AWG Radius Versus Frequency

<sup>\*</sup>The data presented is for sine wave excitation. The author could not find any data for square wave excitation.

#### APPENDIX K

# AREA PRODUCT A RELATIONSHIP

There is a unique relationship between the "Area Product", Ap characteristic number for transformer cores and several other important parameters which must be considered in transformer design.

The power handling capability of a transformer can be related to its  $^{A}p$  quantity (which is actually its  $^{W}a^{A}c$  product where  $^{W}a$  is the available core window area in cm $^{2}$  and  $^{A}c$  is the effective cross-sectional area of the core in cm $^{2}$ ).

These relationships can now be used as new tools to simplify and standardize the process of transformer design. They make it possible to design transformers of smaller bulk and volume or to optimize efficiency.

Table Kl was developed using the least-squares curve fit from the data obtained in Tables K2 through K6. The area product A<sub>p</sub> relationships with volume, surface area, current density, and weight for tape wound cores, C type core, powder cores, laminations and pot core are found in Figures K1 through K20.

Table Kl. Transformer Configuration Constants

|                  | K, 25°C | K <sub>j</sub> 50°C | η     | Ks    | K<br>w | K <sub>v</sub> |
|------------------|---------|---------------------|-------|-------|--------|----------------|
| Pot cores        | 433     | 632                 | -0.17 | 33, 8 | 48.0   | 14,5           |
| Powder cores     | 290     | 423                 | -0.12 | 32, 5 | 58, 8  | 13.1           |
| Lamination       | 366     | 534                 | -0,12 | 41.3  | 68. 2  | 19.7           |
| C type cores     | 323     | 468                 | -0.14 | 39.2  | 66.6   | 17.9           |
| Tape wound cores | 250     | 365                 | -0.13 | 50.9  | 82. 3  | 25.0           |

J = 
$$K_j A_p^{(\eta)}$$
  $A_t = K_s A_p^{0.50}$   
 $W_t = K_w A_p^{0.75}$   $Vol = K_v A_p^{0.75}$ 

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|       | 1            | 2                  | 3                              | 4        | 5                 | 6               | 7      | 8                             | 9                                | 10              | 11    | 12                            | 13                               | 14              | 15                        | 16                             |
|-------|--------------|--------------------|--------------------------------|----------|-------------------|-----------------|--------|-------------------------------|----------------------------------|-----------------|-------|-------------------------------|----------------------------------|-----------------|---------------------------|--------------------------------|
|       | Core         | A <sub>t</sub> cm² | A <sub>p</sub> cm <sup>‡</sup> | MLT cm   | NAWG              | Ω <b>@</b> 50°C | PΣ     | $I = \sqrt{\frac{W}{\Omega}}$ | ΔT 25°C<br>J = 1/cm <sup>2</sup> | Ω <b>@</b> 75°C | Pž    | $1 = \sqrt{\frac{W}{\Omega}}$ | AT 50°C<br>J = I/cm <sup>2</sup> | Total<br>Weight | Volume<br>cm <sup>3</sup> | A <sub>c</sub> cm <sup>2</sup> |
| 1     | 5505)        | 7.19               | 0.0437                         | - 2,12   | 86 25             | 0, 215          | 0.216  | 0, 706                        | 435                              | 0.236           | 0.503 | 1,03                          | 635                              | 5, 81           | 1.39                      | 0.113                          |
| Z     | 55121        | 12.3               | 0.137                          | 2.71     | 160 <sub>25</sub> |                 | 0. 369 | 0.599                         | 369                              | 0, 563          | 0.861 | 0, 874                        | 538                              | 13.3            | 3. 11                     | 0, 196                         |
| 3     | 55848        | 17.3               | 0,259                          | 4. 95    | 257 25            | 0.897           | 0, 519 | 0.537                         | 344                              | 0.985           | 1.211 | 0,783                         | 502                              | 21,3            | 5.07                      | 0. 232                         |
| 4     | 55059        | 21.9               | 0.466                          | 3, 39    | 316 25            | 1.27            | 0,657  | 0.508                         | 314                              | 1.39            | 1.533 | 0, 742                        | 458                              | 32, 3           | 7.28                      | 0, 327                         |
| 5     | 55894        | 30.8               | 1,021                          | 4, 51    | 351 <sub>25</sub> | 1.87            | 0. 924 | 0.496                         | 306                              | 2.06            | 2. 16 | 0, 724                        | 447                              | 59,4            | 12.4                      | 0.639                          |
| 6     | 55586        | 48.6               | 1.821                          | 4.30     | 902 25            |                 | 1.46   | 0.394                         | 244                              | 5, 15           | 3.40  | 0, 574                        | 355                              | 94. 9           | 23.3                      | 0. 458                         |
| 7     | 55071        | 44.7               | 1.966                          | 4.77     | 656 25            | 3, 70           | 1. 34  | 0.425                         | 263                              | 4. 07           | 3, 13 | 0.620                         | 383                              | 94.4            | 21.0                      | 0.666                          |
| В     | 55076        | 51.6               | 2,46                           | 4.88     | 815<br>25         | 4, 71           | 1, 55  | 0, 405                        | 250                              | 5. 17           | 3,61  | 0.590                         | 365                              | 113.0           | 25.7                      | 0,670                          |
| 9     | 55083        | 66.8               | 4,57                           | 6, 02    | 959 25            |                 | 2, 00  | 0,382                         | 236                              | 7. 50           | 4.68  | 0.558                         | 345                              | 178.0           | 39. 1                     | 1,06                           |
| 10    | 55090        | 89.4               | 8,19                           | 6,65     | 1372 25           | 10.8            | 2,68   | 0, 352                        | 225                              | 11.8            | 6.26  | 0,513                         | 329                              | 271.0           | 59.5                      | 1.32                           |
| 11    | 55439        | 86.9               | 8,48                           | 7,48     | 959 25            | 8,40            | 2,60   | 0.391                         | 250                              | 9. 32           | 6,08  | 0.571                         | 365                              | 291.0           | 58. I                     | 1.95                           |
| 12    | 55716        | 100.0              | 9.38                           | 6,54     | 1684 25           | 13,0            | 3,00   | 0.339                         | 217                              | 14.3            | 7.00  | 0, 494                        | 317                              | 303,0           | 69.0                      | 1.24                           |
| 13    | 55110        | 124.0              | 13.66                          | 7. 09    | 2125 25           | 17.8            | 3, 72  | 0. 322                        | 206                              | 19.6            | 8.68  | 0, 470                        | 301                              | 405,0           | 93.4                      | 1,44                           |
| coppe | er loss = ir | on loss            | <u> </u>                       | <u> </u> |                   | ·               |        |                               | <del>*</del>                     |                 |       |                               |                                  |                 |                           |                                |



## Definitions for Table K2

## Information given is listed by column as:

- 1. Manufacturer part number
- 2. Surface area calculated from Figure K21
- 3. Area product effective iron area times window area
- 4. Mean length turn
- 5. Total number of turns and wire size using a window utilization factor  $K_{ij} = 0.40$
- 6. Resistance of the wire at 50°C
- 7. Watts loss is based on Figure Cl for a  $\Delta T$  of 25°C with a room ambient of 25°C surface dissipation times the transformer surface area, total loss is equal to 2  $P_{CU}$
- Current calculated from column 6 and 7
- 9. Current density calculated from column 5 and 8
- 10. Resistance of the wire at 75°C
- 11. Watts loss is based on Figure Cl for a  $\Delta T$  of 50°C with a room ambient of 25°C surface dissipation times the transformer surface area, total loss is equal to 2  $P_{Cll}$
- 12. Current calculated from column 10 and 11
- 13. Current density calculated from column 5 and 12
- 14. Effective core weight for silicon plus copper weight
- 15. Transformer volume calculated from Figure K24
- 16. Core effective cross-section

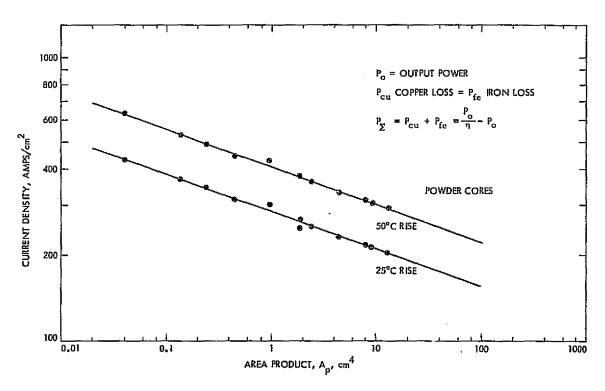


Fig. K1. Current Density Versus Area Product  $A_p$  for a 25°C and 50°C Rise for Powder Cores

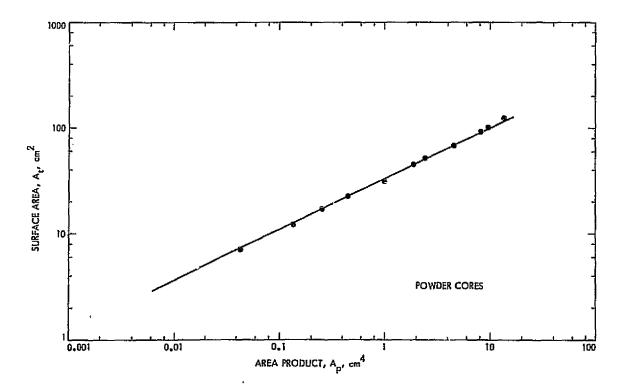


Fig. K2. Surface Area Versus Area Product  $A_{\mathbf{p}}$  for Powder Cores

P

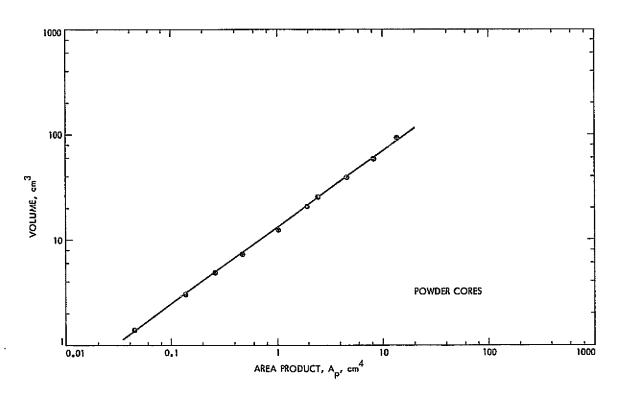


Fig. K3. Volume Versus Area Product  $\mathbf{A}_p$  for Powder Cores

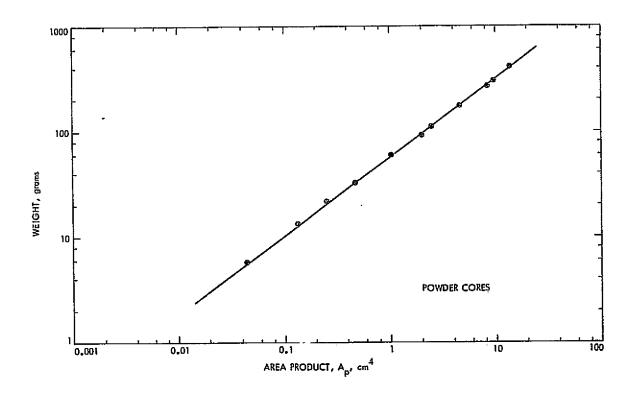


Fig. K4. Total Weight Versus Area Product  $\mathbf{A}_p$  for Powder Cores

|      | 1            | 2                              | 3                              | 4      | 5                | 6        | 7      | 8                             | 9                                | 10       | 11     | 12                  | 13                               | 14              | 15                        | 16                 |
|------|--------------|--------------------------------|--------------------------------|--------|------------------|----------|--------|-------------------------------|----------------------------------|----------|--------|---------------------|----------------------------------|-----------------|---------------------------|--------------------|
|      | Core         | A <sub>t</sub> am <sup>2</sup> | A <sub>p</sub> cm <sup>4</sup> | MLT cm | N AWG            | Ω @ 50°C | PΣ     | $I = \sqrt{\frac{W}{\Omega}}$ | ΔΤ 25°C<br>J = I/cm <sup>2</sup> | Ω @ 75°C | PΣ     | I =√ <del>W</del> Ω | ΔT 50°C<br>J = I/em <sup>2</sup> | Total<br>Weight | Volume<br>cm <sup>3</sup> | A <sub>c</sub> cm² |
| 1    | 9 x 5        | 2.93                           | 0.0065                         | 1,85   | <sup>25</sup> 30 | 0.175    | 0.098  | 0.529                         | 1044                             | 0, 192   | 0.230  | 0.774               | 1527                             | 1.12            | 0.367                     | 0, 10              |
| 2    | 11 x 7       | 4. 35                          | 0.0152                         | 2,2    | 37 30            | 0.309    | 0, 130 | 0.458                         | 904                              | 0. 339   | 0, 304 | 0.670               | 1322                             | 2.08            | 0.662                     | 0.16               |
| 3    | 14 x 8       | 6.96                           | 0.0393                         | 2.8    | <sup>74</sup> 30 |          | 0,208  | 0.363                         | 716                              | 0, 864   | 0,487  | 0.531               | 1048                             | 4.18            | 1.35                      | 0.25               |
| 4    | 18 x 11      | 11. 3                          | 0.114                          | 3, 56  | 143 30           |          | 0, 339 | 0,296                         | 584                              | 2.12     | 0.791  | 0.432               | 853                              | B. 37           | 2,78                      | 0.43               |
| 5    | 22 x 13      | 17.0                           | 0,246                          | 4.4    | Z07 30           |          | 0.510  | 0.271                         | 535                              | 3.80     | 1.190  | 0.396               | 782                              | 17.3            | 5.17                      | 0,63               |
| 6    | 26 × 16      | 23.9                           | 0.498                          | 5,2    | 96 <sub>25</sub> | 0.592    | 0.717  | 0.778                         | 479                              | 0.650    | 1.67   | 1.13                | . 696                            | 28. 5           | 8.65                      | 0.94               |
| 7    | 30 × 19      | 32.8                           | 1.016                          | 6.0    | 144 25           | 1.024    | 0.984  | 0.693                         | 427                              | 1, 12    | 2.30   | 1.01                | 559                              | 48.9            | 13.9                      | 1.36               |
| 8    | 36 × 22      | 44.8                           | 2.01                           | 7.3    | 189 25           | 1.636    | 1.34   | 0.639                         | 394                              | 1,79     | 3. 14  | 0.937               | 577                              | 77, 8           | 0.55                      | 2.01               |
| 9    | 47 x 28      | 76,0                           | 5.62                           | 9.3    | 345 25           | 3.81     | 2, 28  | 0.547                         | 337                              | 4. 18    | 5. 32  | 0.798               | 492                              | 173.0           | 48.6                      | 3.12               |
| 10   | 59 × 36      | 122.0                          | 13.4                           | 12.0   | 608 25           | 8.65     | 3. 66  | C. 459                        | 263                              | 9,50     | 8.54   | 0.670               | 413                              | 379, 0          | 98.3                      | 4,85               |
| сорр | er loss = ir | n loss                         | <u> </u>                       |        |                  |          |        |                               |                                  |          |        |                     |                                  |                 |                           |                    |

Table K3. Pot cores characteristics

OF POOR QUALITY

# Definition for Table K3

- 1. Manufacturer part number
- 2. Surface area calculated from Figure K21
- 3. Area product effective iron area times window area
- 4. Mean length turn
- 5. Total number of turns and wire size using a window utilization factor  $K_{ij} = 0.40$
- 6. Resistance of the wire at 50°C
- 7. Watts loss is based on Figure Cl for a  $\Delta T$  of 25°C with a room ambient of 25°C surface dissipation times the transformer surface area, total loss is equal to 2  $P_{CU}$
- 8. Current calculated from column 6 and 7
- 9. Current density calculated from column 5 and 8
- 10. Resistance of the wire at 75°C
- 11. Watts loss is based on Figure Cl for a  $\Delta T$  of 50°C with a room ambient of 25°C surface dissipation times the transformer surface area, total loss is equal to 2  $P_{cu}$
- . 12. Current calculated from column 10 and 11
  - 13. Current density calculated from column 5 and 12
  - 14. Effective core weight for silicon plus copper weight
  - 15. Transformer volume calculated from Figure K24
  - 16. Core effective cross-section

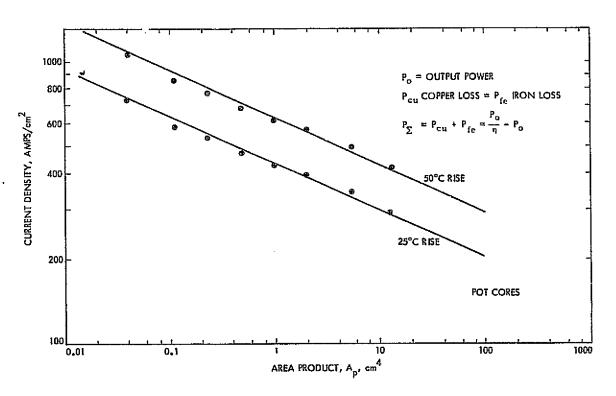


Fig. K5. Current Density Versus Area Product  $A_p$  for a 25°C and 50°C Rise for Pot Cores

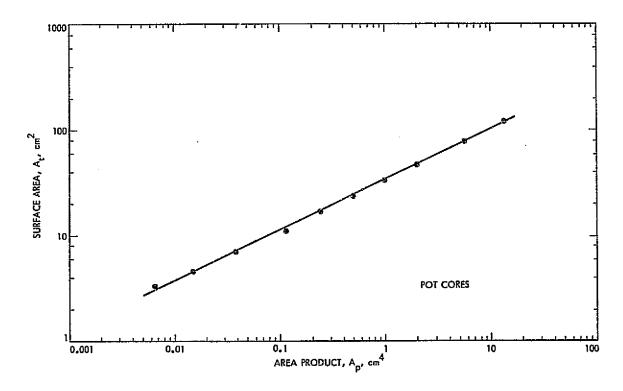


Fig. K6. Surface Area Versus Area Product  $A_p$  for Pot Cores

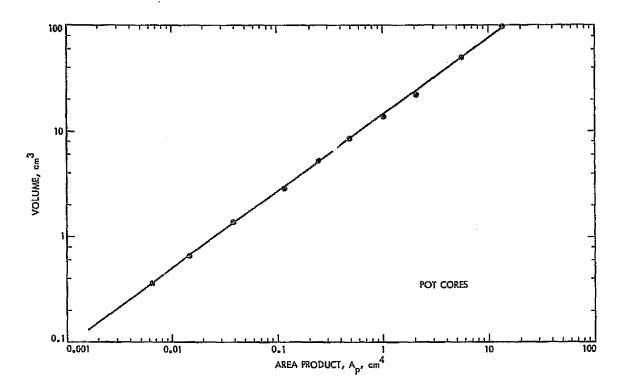


Fig. K7. Volume Versus Area Product  $A_p$  for Pot Cores

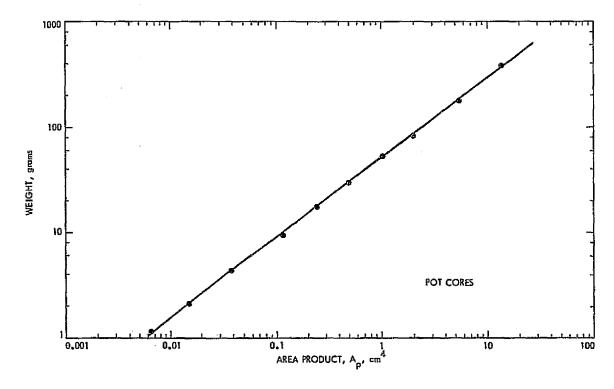


Fig. K8. Total Weight Versus Area Product  $\mathbf{A}_{\mathbf{p}}$  for Pot Cores

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|    | 1       | 2                              | 3                              | 4      | 5                 | 6               | 7     | 8  | 9                                | 10      | 11             | 12                            | 13                               | 14              | 15                        | 16                 |
|----|---------|--------------------------------|--------------------------------|--------|-------------------|-----------------|-------|--|----------------------------------|---------|----------------|-------------------------------|----------------------------------|-----------------|---------------------------|--------------------|
|    | Core    | A <sub>t</sub> cm <sup>2</sup> | A <sub>p</sub> cm <sup>4</sup> | MLT cm | NAWG              | Ω <b>€</b> 50°C | Pr    | $I = \sqrt{\frac{\overline{W}}{\Omega}}$ | ΔT 25°C<br>J = 1/cm <sup>2</sup> | ถ €75°C | P <sub>2</sub> | 1 # $\sqrt{\frac{w}{\Omega}}$ | AT 50°C<br>J = 1/cm <sup>2</sup> | Total<br>Weight | Volume<br>cm <sup>3</sup> | A <sub>c</sub> cm² |
| 1  | EE-3031 | 4.11                           | 0.0090                         | 1.72   | 90 30             | 0,58            | 0.123 | 0, 323                                   | 638                              | 0.645   | 0.288          | 0.472                         | 932                              | 2, 04           | 0.651                     | 0.056              |
| 2  | EE-2829 | 6.63                           | 0. 0254                        | 2,33   | 147 30            | 1,/30           | 0.199 | 0, 276                                   | 546                              | 1, 43   | 0.464          | 0,403                         | 795                              | 3.78            | 1.35                      | 0, 101             |
| 3  | E1-187  | 14.4                           | 0.120                          | 3,20   | 314 30            | 3. 82           | 0.432 | 0.237                                    | 469                              | 4.19    | 1.01           | 0.347                         | 685                              | 10,2            | 4, 34                     | 0,225              |
| 4  | EE-2425 | 23. B                          | 0, 325                         | 5. OB  | 498 30            | 9.61            | 0,714 | 0, 192                                   | 380                              | 10.5    | 1.67           | 0.281                         | 555                              | 24,6            | 9.22                      | 0, 403             |
| 5  | EE-2627 | 40.6                           | 1.01                           | 5.79   | 245 25            | 1.68            | 1.22  | 0.602                                    | 371                              | 1.85    | 2.84           | 0.876                         | 540                              | 61.3.           | 19. 1                     | 0.907              |
| 6  | E1-375  | 47.7                           | 1.38                           | 6.30   | 350 <sub>25</sub> | 2.62            | 1.43  | 0. 522                                   | 322                              | 2,87    | 3, 34          | 0, 762                        | 470                              | 74.4            | 25, 3                     | 0.907              |
| 7  | E1-50   | 57.7                           | 1.95                           | 7.00   | 263 25            | 2.21            | 1.73  | 0.625                                    | 385                              | 2.43    | 4.04           | 0, 912                        | 562                              | 124.0           | 36. E                     | 1.61               |
| 8  | E1-21   | 66,0                           | 2.62                           | 7, 57  | 372 25            | 3.34            | 1.98  | 0.544                                    | 335                              | 3,66    | 4.62           | 0.793                         | 489                              | 140.0           | 39.2                      | 1, 61              |
| 9  | E1-625  | 90, 0                          | 4, 76                          | 8, 84  | 503 25            | 5, 27           | 2. 70 | 0, 505                                   | 312                              | 5, 79   | 6. 30          | 0. 737                        | 455                              | 223, 0          | 60,0                      | 2, 52              |
| 10 | E1+75   | 130.0                          | 9.87                           | 10.6   | 211 20            | 0.826           | 3.90  | 1,54                                     | 296                              | 0.906   | 9, 10          | 2,24                          | 432                              | 417.0           | 104.0                     | 3, 63              |
| 1  | E1-87   | 176.0                          | 18.3                           | 12.3   | 296 20            | 1,34            | 5. 28 | 1,40                                     | 270                              | 1.48    | 12.3           | 2,04                          | 393                              | 616.0           | 164.0                     | 4. 94              |
| 12 | E1-100  | 230.0                          | 31.2                           | 14,5   | 386 20            | 2,07            | 6.90  | 1.29                                     | 249                              | 2,27    | 16, 1          | 1, 66                         | 363                              | 953.0           | 246.0                     | 6, 45              |
| 13 | E1-112  | 292.0                          | 49. 9                          | 16.0   | 492 ZO            | 2. 91           | 8.76  | 1.23                                     | 237                              | 3, 19   | 20.4           | 1.79                          | 344                              | 1370,0          | 350.0                     | 8. 16              |
| 14 | E1-125  | 361.0                          | 76.3                           | 17.7   | 625 20            | 4,09            | 10.8  | 1, 15                                    | 222                              | 4,40    | 25. 3          | 1.68                          | 324                              | 1870.0          | 481.0                     | 10,08              |
| 15 | E1-138  | 432, 0                         | 112.0                          | 19.5   | 740 20            | 5, 33           | 13.0  | 1, 10                                    | 213                              | 5.85    | 30.2           | 1.61                          | 310                              | 2560.0          | 629.0                     | 12, 19             |
| 16 | E1-150  | 518.0                          | 158.0                          | 21.2   | 893 20            | 6,99            | 15.5  | 1.05                                     | 203                              | 7.67    | 36, 3          | 1.54                          | 296                              | 3360,0          | 829.0                     | 14, 51             |
| 17 | E1-175  | 704.0                          | 292.0                          | 24.7   | 1080 20           | 9,85            | 21.1  | 1,034                                    | 199                              | 10.8    | 49.3           | 1,51                          | 291                              | 5180.0          | 1312.0                    | 19.75              |
| 18 | E1-36   | 778,0                          | 361.0                          | 26,5   | 1701 20           | 16, 6           | 23.3  | 0. 836                                   | i<br>, 161                       | 18.3    | 54, 5          | 1, 22                         | 235                              | 5930.0          | 1654.0                    | 17. 03             |
| 19 | E1-19   | 1093.0                         | 668.0                          | 31.7   | 2006 20           | 33, 8           | 32. 8 | 0,696                                    | 134                              | 37.1    | 76, 5          | 1.015                         | 196                              | 8694, 0         | 2875.0                    | 19.75              |



# Definitions for Table K4

- 1. Manufacturer part number
- 2. Surface area calculated from Figure K22
- 3. Area product effective iron area times window area
- 4. Mean length turn on one bobbin
- 5. Total number of turns and wire size for one bobbin using a window utilization factor  $K_{IJ} = 0.40$
- 6. Resistance of the wire at 50°C
- 7. Watts loss is based on Figure Cl for a  $\Delta T$  of 25°C with a room ambient of 25°C surface dissipation times the transformer surface area, total loss is equal to 2  $P_{Cl}$
- 8. Current calculated from column 6 and 7
- 9. Current density calculated from column 5 and 8
- 10. Resistance of the wire at 75°C
- 11. Watts loss is based on Figure Cl for a  $\Delta T$  of 50°C with a room ambient of 25°C surface dissipation times the transformer surface area, total loss is equal to 2  $P_{Cl}$
- 12. Current calculated from column 10 and 11
- 13. Current density calculated from column 5 and 12
- 14. Effective core weight for silicon plus copper weight
- 15. Transformer volume calculated from Figure K25
- 16. Core effective cross-section

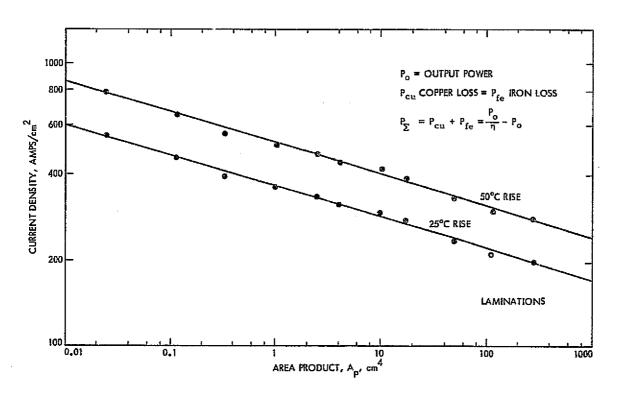


Fig. K9. Current Density Versus Area Product Ap for 25°C and 50°C Rise for Laminations

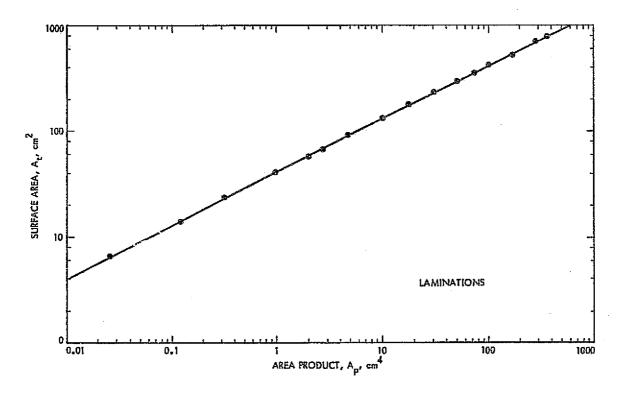


Fig. K10. Surface Area Versus Area Product  $\mathbf{A}_{\mathbf{p}}$  for Laminations

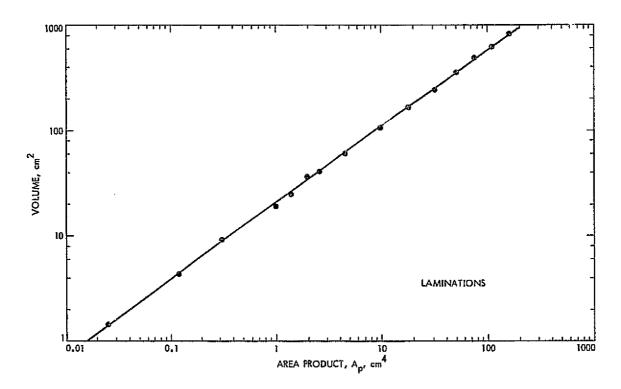


Fig. K11. Volume Versus Area Product  $\mathbf{A}_{\mathbf{p}}$  for Laminations

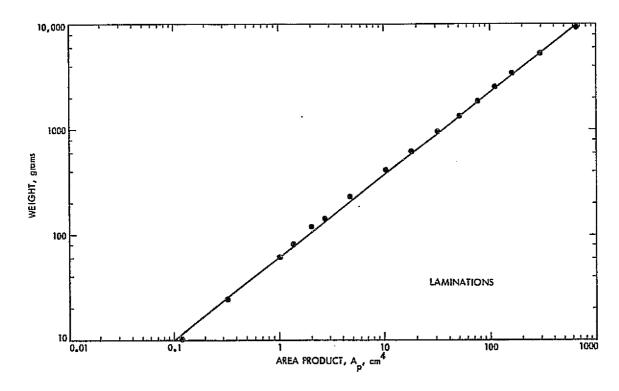


Fig. K12. Total Weight Versus Area Product  $\mathbf{A}_p$  for Laminations

| 89 to   10 to   160   167   16 to   95 to   16 to   167   16 to   95 to   16 to   167   16 to   16 t    | 91                   | ST             | 74          | ET           | 15           | tī    | ot             | 6           | 8            | L                 | 9             | Ē                 | +             | €                   | Z                                 | Ţ,              | ] |
|---|----------------------|----------------|-------------|--------------|--------------|-------|----------------|-------------|--------------|-------------------|---------------|-------------------|---------------|---------------------|-----------------------------------|-----------------|---|
| 85 to   167   16    | A <sub>cetta</sub> 2 | emuloV<br>fmp  | MaieW InfoT |              | <u>w</u> √.1 | z d   | D •€1 ⊕ #      |             | <u>#</u> ∧:1 | χ <sub>d</sub>    | ე .05 A ც     | DAV N             | ητο T.J.M.    | t <sub>mo q</sub> A | 2 <sup>tti 2</sup> <sup>1</sup> V | aren            |   |
| 89   6   167   75   161   6   6   75   161   6   6   75   161   6   75   161   6   75   75   161   6   75   75   161   6   75   75   161   6   75   75   161   6   75   75   161   6   75   75   75   75   75   75   75   | 295.0                | 1,14           | £5.63       | 87S          | £75.0        | 91 I  | t# *6          | 07£         | 7H1.0        | 753.0             | €6 'H         | 0f 299            | ÷÷ .!         | £9 <b>7 '</b> 0     | 9,05                              | 2-7 <b>V</b>    |   |
| SP   1  | 011-19               | 26 'R          | 81"16       | 775          | 697 *0       | 49°T  | 6.11           | \$9E        | 981'0        | 7 17 ,0           | 10' 2         |                   | 41,18         | 014.0               | 6,15                              | £-7¥            |   |
| 99   10   1945   0   1966   167   16   1   19   1   19   1   19   1   19   1   1  | 655 0                | 90 °F1         | 8*ts        | £6F          | \$\$7.0      | ₹£,5  | t*xt           | 949         | F41.0        | το'τ              | ē.01          | OF OP6            | 6 <u>5</u> 't | 137.0               | 2.88                              | 5-1 <b>∀</b>    |   |
| BP  | 917,0                | RR '≎1         | 1 *59       | 6Ht          | £57 10       | £9'2  | 3.05           | 11-6        | 571.0        | 1,13              | H.81          | 1 1               | £5.è          | 1 10 1              | F.78                              | "- <b>⊤</b> ¥   |   |
| 89   1-   1-   1-   1-   1-   1-   1-   1   | 917.0                | 05 ,55         | 8.08        | EFF          | 655.0        | 11.6  | \$.06          | ott         | 25 t °0      | 9£ '1             | 6.75          |                   | 0÷ '÷         | ++.1                | £.ē4                              | +21-7 <b>V</b>  |   |
| ## 1  | 709°0                | 99.26          | 28.751      | 565          | \$0 '7       | PF 'F | 625 '0         | 147         | 1.404        | 06*1              | 5H+ .0        | 02 125            | F7 ,č         | 15 7                | F*£'9                             | 8-1V            |   |
| ## 1  | 1,077                | 29 ' <b>19</b> | 8*551       | 165          | 10'7         | £R'+  | Z#5 "0         | 892         | 6t.£         | 70.5              | €E± 10        | 1 1               | HE *7         | 4.0 1€              | 0.69                              | 6-7 <b>V</b>    | 1 |
| 99 to   1   | 1, 342               | 55.74          | 183.2       | 78E          | tu ,5        | 55,22 | 9 <b>+9</b> *0 | 99 <b>2</b> | 86.1         | +2.5              | HH2 '0        | 1 1               | 10.7          | SR E                | ⊆ * <b>‡</b> 4                    | ¥F-10           |   |
| #E.I. 28.50   0.882   472   494.1   78.30   510.0   365   26.11   96.2   118.0   0.5 \$12   10.7   10.00   1.89   1.000 | 97 °T                | 86 ° 19        | 2,405       | 146          | £6.1         | 60*9  | 158,0          | 562         | 58.2         | 13.5              | HF1 -0        | 05 875            | 60 '4         | 25.75               | 0.78                              | \$1-3 <b>V</b>  |   |
| 57'I     66'BO     07125     98E     60'I     97'I     97'I     98'I   | 1, 26                | £9.69          | 0.755       | 5 <b>†</b> f | 14.1         | 95.0  | 769,0          | 0+7         | +5.1         | 18.5              | 806,0         | 05 656            | 3E.7          | +1 '5               | 7.10                              | 4£1-13¥         |   |
| 68***     66***     65***   | 7° 34                | ₹ <b>Я</b> *79 | 0 '857      | ተረና          | #6 °I        | 2819  | 316.0          | 952         | ££.1         | 46.5              | \$1 M .O      |                   | 10 4          | ,n.a                | 1 '86                             | HZ-7V           |   |
| 89.4     0.490     0.490     0.400     <  | 57 '[                | 64,74          | 0.156       | 205          | 1. 60        | 92°H  | 1911           | 117         | 01-1         | ₹ċ€               | 46 ° I        | 07 <sub>D15</sub> | 1914          | 74. 1               | ×ft                               | WT-114          |   |
| ## ## ## ## ## ## ## ## ## ## ## ## ##  | e# t                 | £4.49          | 0 '750      | 945          | 6£*I         | 0F.8  | 06.1           | 765         | 1.23         | ₩ <b>⊊</b> *€     | я <b>1</b> 't | 986 20            | £0 'R         | ₹u.•                | 170                               | \$T-7 <b>V</b>  |   |
| 89' 15  | ŀ                    | ! !            |             |              |              | 68.8  | 1.43           | EFZ         | 07.1         | 08.8              | 0f 't         |                   | 6H 'H         | 8'01                | 75.1                              | 91-7♥           |   |
| 85'E 80'L81 0'512 016 T67 15'1 1'51 25'5 007 950'1 09'9 20'5 07 25'9 2'11 0'87 707 77'1W  85'E 80'L81 0'512 016 10'1 2'11 2'11 55'7 512 90'1 95'5 57'7 07 25'9 2'11 0'87 707 77'1W  | 1                    | ] [            |             |              |              | F6'6  | 99*1           | 827         | 681"1        | 4.25              | Is 'I         | 1 1               | £ '01         | + '+1               | . 241                             | 41-7 <b>V</b>   |   |
| 85 to 0.515 0.868 E05 22.1 1.01 70.2 005 80.1 00.0 1.05 00.1 00.0 70.1 0.00 70.1 0.00 70.1 0.00 25.1 0.00   |                      | 1 1            |             |              |              |       |                |             | 640°t        | £2.1 <del>1</del> | 01.5          |                   | н 'от         | .81                 | 651                               | 61-7 <b>V</b>   |   |
| 89 % La 1995 0 1966 167 15 1 15 1 251 25 002 950 1 09 9 10 1 07 15 07 1 6 1 6 1 6 1 6 1 6 1 6 1 6 1 6 1 6 1   |                      | 1 1            | 1           |              |              |       |                |             | 1            |                   | ĺ             |                   |               | 1.22                | 741                               | 115-41 <b>∧</b> |   |
|   | l                    | 1              |             |              |              |       |                | ·           | 1            |                   |               |                   |               | •                   |                                   | 27-1V           |   |
| 89'E 16'097 0'001 692 6E'T 1'21 +2'+ 821 226'O 5E'2 7E'+ 0'20 0'21 0'00 5E'2 7E'-TV   | 95 '£                | 16 '087        | 0 '0601     | 69Z          | 1" 32        | 1.71  |                |             | i.           |                   | i             | 02 844<br>02 444  |               | · -                 |                                   | +2-7V           |   |

#### Definitions for Table K5

- 1. Manufacturer part number
- 2. Surface area calculated from Figure K23
- 3. Area product effective iron area times window area
- 4. Mean length turn on one bobbin
- 5. Total number of turns and wire size for two bobbins using a window utilization factor  $K_{ii}$  = 0.40
- 6. Resistance of the wire at 50°C
- 7. Watts loss is based on Figure Cl for a  $\Delta T$  of 25°C with a room ambient of 25°C surface dissipation times the transformer surface area, total loss is equal to 2  $P_{cu}$
- 8. Current calculated from column 6 and 7
- 9. Current density calculated from column 5 and 8
- 10. Resistance of the wire at 75°C
- 11. Watts loss is based on Figure Cl for a  $\Delta T$  of 50°C with a room ambient of 25°C surface dissipation times the transformer surface area, total loss is equal to 2  $P_{Cl}$
- 12. Current calculated from column 10 and 11
- 13. Current density calculated from column 5 and 12
- 14. Effective core weight for silicon plus copper weight
- 15. Transformer volume calculated from Figure K26
- 16. Core effective cross-section

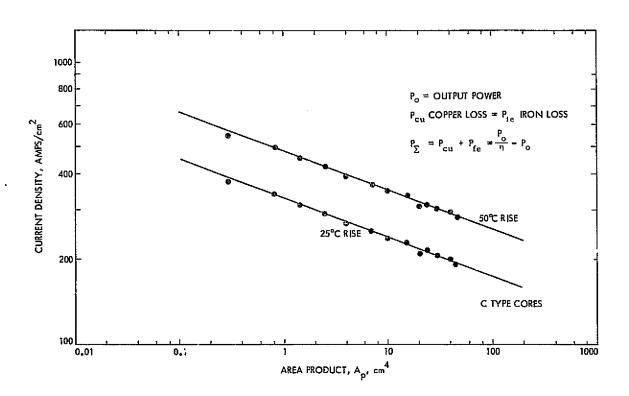


Fig. K13. Current Density Versus Area Product  $A_p$  for 25°C and 50°C Rise for C-Type Cores

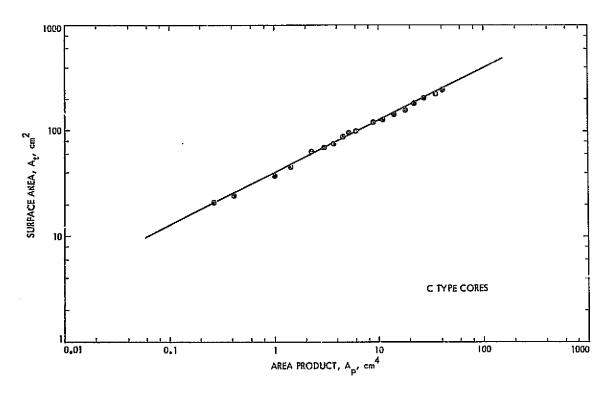


Fig. K14. Surface Area Versus Area Product Ap for C-Type Cores

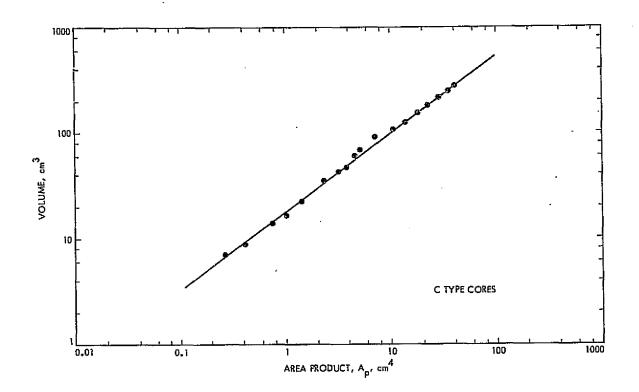


Fig. K15. Volume Versus Area Product  $\mathbf{A}_{\mathbf{p}}$  for C-Type Cores

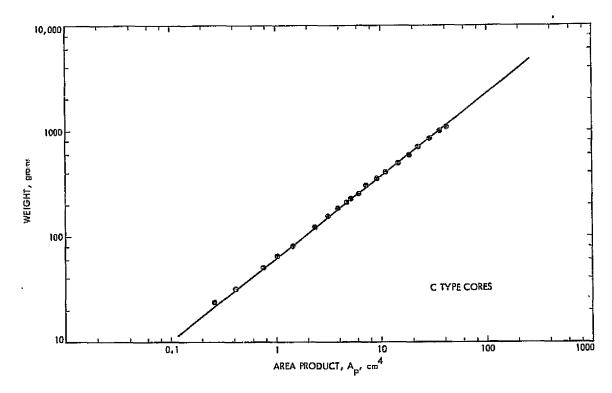


Fig. K16. Total Weight Versus Area Product  $\mathbf{A}_p$  for C-Type Cores

|           | 1      | 2                              | 3                  | 4                | 5                  | 6       | 7      | 8                             | 9                                | 10     | 11            | 12                                       | 13                               | 14              | 15                        | 16                             | ]        |
|-----------|--------|--------------------------------|--------------------|------------------|--------------------|---------|--------|-------------------------------|----------------------------------|--------|---------------|--|----------------------------------|-----------------|---------------------------|--------------------------------|----------|
|           | Core   | A <sub>t</sub> cm <sup>2</sup> | Ap cm <sup>3</sup> | MLT cm           | N AWG              | Ω∰50°C  | PI     | $1 = \sqrt{\frac{W}{\Omega}}$ | ΔΤ 25°C<br>J = I/cm <sup>2</sup> | ถ@75°C | PΣ            | $I = \sqrt{\frac{\overline{W}}{\Omega}}$ | Δ7 50°C<br>J = 1/cm <sup>2</sup> | Total<br>Weight | Volume<br>cm <sup>3</sup> | A <sub>c</sub> cm <sup>2</sup> |          |
| 1         | 52402  | 7,26                           | 0,0100             | 2,05             | 302 <sub>30</sub>  | 2, 35   | 0.218  | 0,215                         | 425                              | 2.58   | 0, 508        | 0.313                                    | 619                              | 3, 75           | 1, 42                     | 0.022                          | ]        |
| 2         | 52153  | 8.29                           | 0.0176             | 2.22             | 302 30             | 2.54    | 0.249  | 0,221                         | 436                              | 2.80   | 0,580         | • 0.32≥                                  | 636                              | 4,60            | 1.71                      | 0.053                          |          |
| 3         | 52107  | 11, 1                          | 0.0201             | 2, 21            | 606 30             | 5, 09   | 0.333  | 081.0                         | 357                              | 5, 59  | 0.777         | 0, 263                                   | 520                              | 7.64            | 2.63                      | 0. 022                         | ; ;      |
| 4         | 52403  | 13.5                           | 9. 0267            | 2.30             | 621 30             | 5,43    | 0,405  | 0.193                         | 381                              | 5, 96  | 0.945         | 0.281                                    | 556                              | 10.4            | 3.48                      | 0, 022                         | Tabl     |
| 5         | 52057  | 17.4                           | 0,0659             | 2.53             | 1017 30            | 9.78    | 0. 522 | 0.163                         | 342                              | 10.7   | 1,22          | 0.238                                    | 471                              | 15.1            | 4.98                      | 0, 043                         | O        |
| 6         | 52000  | 15.2                           | 0.0787             | 2.70             | 606 30             | 5.22    | 0.456  | 0, 191                        | 378                              | 6.82   | 1,00          | 0.278                                    | 550                              | 11.7            | 3.99                      | 0, 086                         | 5        |
| 7         | 52063  | 20.7                           | 0.132              | 2.85             | 1017 30            | 11.0    | 0,621  | 0, 167                        | 331                              | 12.1   | 1.45          | 0. 244                                   | 483                              | 18.9            | 6.20                      | 0.086                          | ١٠,      |
| 8         | 52002  | 23.8                           | 0.144              | 2,88             | 1114 30            | 12.2    | 0,654  | 0, 163                        | 323                              | 13.4   | 1.53          | 0.239                                    | 472                              | 20.6            | 6.72                      | 0.086                          | H        |
| 9         | 52007  | 27.6                           | 0,380              | 3, 87            | 982 30             | 14.4    | 0.828  | 0.169                         | 334                              | 15,8   | 1.93          | 0.246                                    | 487                              | 32.2            | 9.84                      | 0, 257                         | <u> </u> |
| 10        | 52167  | 31.5                           | 0.516              | 4. 23            | 1000 30            | 16. 1   | 0. 945 | 0, 171                        | 338                              | 17.6   | 2.21          | 0.250                                    | 494                              | 39, 9           | 11. 9                     | 0, 343                         | ape      |
| 11        | 52094  | 30.4                           | 0, 592             | 4. 47            | 1017 30            | 17.3    | 0.912  | 0. 162                        | 321                              | 19.0   | 2.13          | 0.237                                    | 468                              | 42.8            | 12,2                      | 0, 386                         | 8        |
| 12        | 52004  | #6. I                          | 0.725              | 4. OZ            | 315 20             | 0.469   | 1.38   | 1,20                          | 234                              | 0.515  | 3.23          | 1.77                                     | 341                              | 70, 2           | 21, 3                     | 0.171                          | wound    |
| 13        | 52032  | 56.5                           | 1.46               | 4.65             | 315 20             | 0.543   | 1.69   | 1,25                          | 240                              | 0.596  | 3. 95         | 1.82                                     | 351                              | 93, 5           | 27, B                     | 0. 343                         | ם        |
| 14        | 52026  | 61.0                           | 2.18               | 5, 28            | 315 20             | 0,616   | 1, 63  | 1.22                          | 235                              | 0.676  | 4.27          | 1.77                                     | 342                              | 116.0           | 32, 8                     | 0.514                          |          |
| 13        | 52038  | 65.4                           | 2.91               | 5, 97            | 315 20             | 0. 0.17 | 1.98   | 1.19                          | 230                              | 0.765  | 4,61          | 1.74                                     | 334 .                            | 139.0           | 38, 3                     | 0,686                          | Cor      |
| 16        | 52035  | 89.9                           | 4, 68              | 6.33             | 505 20             | 1. 14   | 2.67   | 1.06                          | 204                              | 1.3    | 6. 22         | 1.55                                     | 298                              | 210.0           | 59. D                     | 0,686                          | 0        |
| 17        | 52055  | 116.0                          | 6.81               | 6. 76            | 737 20             | 1. 85   | 3, 48  | 0, 970                        | 187                              | 2.0    | 8. 12         | 1.42                                     | 273                              | 303.0           | 86.4                      | 0, 686                         | chara    |
| 18        | 52012  | 110.0                          | . 4, 35            | 6.8 <sub>B</sub> | 505 20             | 1. 00   | 3, 30  | 0, 496                        | 192                              | 1, 92  | 7.70          | 1,45                                     | 280                              | 378, Q          | B7.4                      | 1. 371                         | l ti     |
| 19        | 521417 | 17%.0                          | 12.5.              | 7.51             | 69 <b>8</b> 17     | 0,97    | 5.37   | 1.66                          | 160                              | 1.065  | 12.5          | 2.33                                     | 274                              | 562.0           | 163.0                     | 0,686                          | 8        |
| 50        | 52031  | 256.0                          | 19,8               | 8, 23            | 1114 17            | 1, 70   | 7,68   | 1,50                          | 145                              | 1,86   | 17. 9         | 2.19                                     | 211                              | 431.0           | 272.0                     | 0.686                          | Cte      |
| 41        | 52103  | 220.0                          | 24.5               | 8,77             | 688 17             | 1.12    | 6.60   | 1. 72                         | 165                              | 1.23   | 15, 4         | ⊒,51                                     | 241                              | 741.0           | 212.0                     | 1, 371                         | 13.      |
| 74        | 54148  | 304, 0                         | 39.4               | 9.49             | 1104 <sub>17</sub> | 1.04    | 9.12   | 1,53                          | 147                              | 2.13   | 21.3          | 2,24                                     | £15                              | 1182.0          | 341.0                     | 1,371                          | stic     |
| 43        | 54024  | .256. C                        | 49. 1              | 11, 3            | 688 17             | 1. 44   | 7.68   | 1.63                          | 157                              | 1,58   | <u>1</u> 7, q | 2,38                                     | 229                              | 1106.0          | 291.0                     | 2,742                          | S        |
| 24        | 37,042 | 347.0                          | 78.7               | 12.0             | 1104 17            | 2. 45   | 10. 4  | 1. 45                         | 140                              | 2, 69  | 24, 3         | 2. 12                                    | 204                              | 1681.0          | 453 <b>.</b> 0            | 2, 742                         | "        |
| 25        | 32100  | 422.0                          | 145, 0             | 15.4             | 1089 17            | 3.11    | 12.7   | 1, 43                         | 138                              | 3,41   | 29.5          | 2.08                                     | 200                              | 2459.0          | 633.0                     | 5. 142                         | ľ        |
| 26        | 32112  | 37B. 0                         | 510.0              | 20.3             | 2871 17            | 10.8    | 26.3   | 1.1                           | 106                              | 11.8   | 61.5          | 1,61                                     | 155                              | 7100.0          | 1891.0                    | 6,855                          | ĺ        |
| <b>17</b> | 12424  | 1014.0                         | 813.0              | 22, 2            | 2856 17            | 11.7    | 24.4   | 1,02                          | 98.1                             | 12.9   | 71.0          | 1.66                                     | 159                              | 8891.0          | 2299,0                    | 10, 968                        |          |

JPL Technical Memorandum 33-767 conser loss - cros loss

### Definitions for Table K6

- 1. Manufacturer part number
- 2. Surface area calculated from Figure K21
- 3. Area product effective iron area times window area
- 4. Mean length turn
- 5. Total number of turns and wire size using a window utilization factor  $K_{11} = 0.40$
- 6. Resistance of the wire at 50°C
- 7. Watts loss is based on Figure Cl for a  $\Delta T$  of 25°C with a room ambient of 25°C surface dissipation times the transformer surface area, total loss is equal to 2  $P_{Cl}$
- 8. Current calculated from column 6 and 7
- 9. Current density calculated from column 5 and 8
- 10. Resistance of the wire at 75°C
- 11. Watts loss is based on Figure Cl for a  $\Delta T$  of 50°C with a room ambient of 25°C surface dissipation times the transformer surface area, total loss is equal to 2  $P_{CH}$
- 12. Current calculated from column 10 and 11
- 13. Current density calculated from column 5 and 12
- 14. Effective core weight plus copper weight
- 15. Transformer volume calculated from Figure K24
- 16. Core effective cross-section

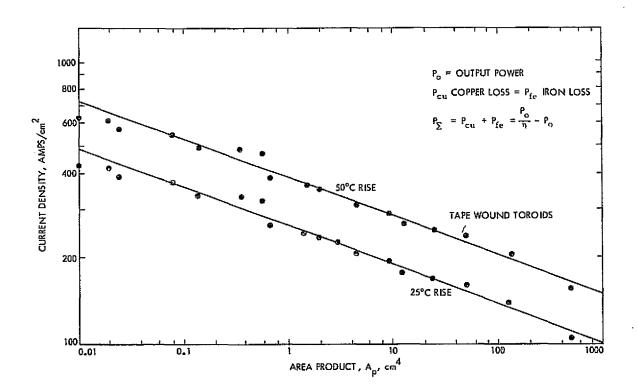


Fig. K17. Current Density Versus Area Product A  $_p$  for 25°C and 50°C Rise for Tape-Wound Toroids

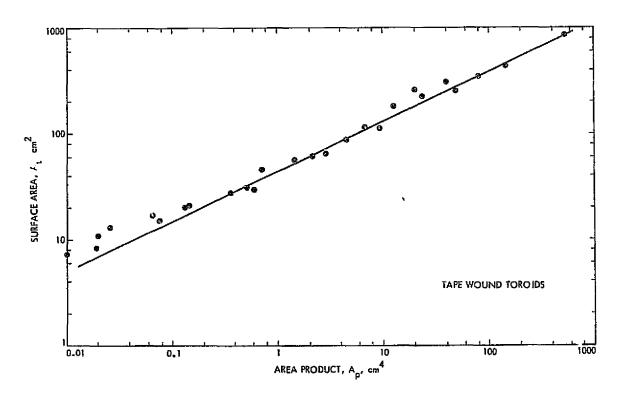


Fig. K18. Surface Area Versus Area Product Ap for Tape-Wound Toroids

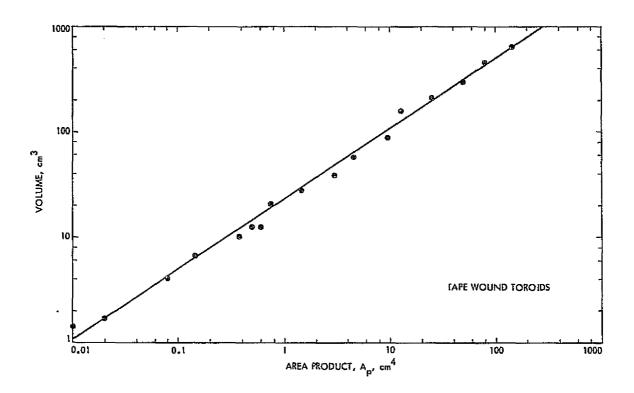


Fig. K19. Volume Versus Area Product  $\mathbf{A}_{\mathbf{p}}$  for Tape-Wound Toroids

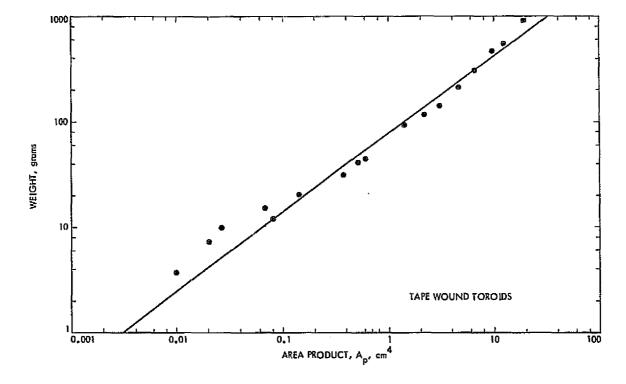


Fig. K20. Total Weight Versus Area Product Ap for Tape-Wound Toroids

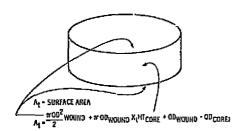


Fig. K21. Tape Wound Core, Power Cores, and Pot Cores Surface Area  $\mathbf{A}_{\mathbf{t}}$ 

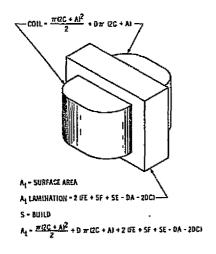


Fig. K22. Lamination Surface Area  $A_t$ 

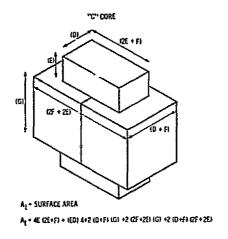


Fig. K23. C Core Surface Area A

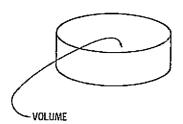


Fig. K24. Tape wound core, Powder Core, and Pot Cores

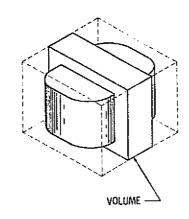


Fig. K25. Laminations

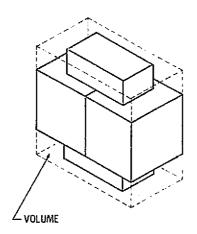


Fig. K26. C Type Cores