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## **SOLID ROCKET PROPELLANT WASTE DISPOSALI INGREDIENT RECOVERY cn I nv (k&SA-CR-l98195) SCLfC ROCKET PBOPELLABT 1176-26351 WASTE DISPCSAL/INGREDIENT RECCVERY STUDY**

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# **FINAL REPORT JPL CONTRACT 954161A**

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**A DIVISION OF THIOKOL CORPORATION** 

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**MER-1374A** 

### SOLID ROCKET PROPELLANT WASTE DISPOSAL/INGREDIENT RECOVERY STUDY

## FINAL REPOFT

JPL CONTRACT 954161A

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#### **I. INTRODUCTION AND SUMMARY**

There exists the technological feasibility, economic and energy conservation incentives, as well **as** the ecological appeal, to recover ingredients from waste propellant. The presence of a large amount of uniform composition waste propellant lends itself to the design and construction of a plant to recover ingredients in a volume great enough to pay out the investment, as well as eliminate the open burning of waste propellant.

This study was initiated by the Jet Propulsion Laboratory **(JPL)** with the aim of better definition of the economic and energy related aspects of waste rocket propellant disposal. This document constitutes the final report as required by the **JPL**  contract.

Environmental and economic pressures are building toward the elimination of open burning as a viable method of waste propellant disposal. The selection, development and implementation of an acceptable alternate is of vital concern. The intelligent selection of the right method is dependent, at least in part, on the economic and energy impact.

**A** comparison of facility and operating costs of the alternate methods shows open burning to be the lowest cost incineration method of waste propellant disposal.

The recovery of ingredients from waste propellant has the probability of being able to pay its way, and even show a profit, when large consistent quantities of composite propellant are available. Ingredients recovered from Space Shuttle waste propellant would be worth over \$1.5 millions,

Open burning and controlled burning are both energy wasteful.

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Ingredlent recovery could conserve **100** million kwh **of** energy on the Space Shuttle program.

#### 11. COST EVALUATIONS OF WASTE PROPELLANT DISPOSAL METHODS

#### A. OPEN BURNING METHOD

In the determination of the costs involved in open burning of waste propellant, use was made of information available at Thiokol.

The replacement cost of an existing burning area has been defined as follows:



The operational manhours per year for one year is:



At a labor rate of \$12 per hour, these manhours translate to a cost of **\$76,800** per year. With the addition of material (such as igniters, gloves, containers, etc. ) costs of **\$1,250** per year and truck rental costs of **\$3,000** per year, the total operating cost for an open burning facility amounts to **\$81,052** per year.

The actual operating cost per pound of propellant burned has been calculated for the last three years as follows:



In contrast to this relatively low cost per pound, one propellant producer in California pays **\$1.00** per **pound** to have his propellant packaged, transported, and burned. The main reasons for this difference are the restrictions placed on open burning in California and the small quantities of propellant involved.

#### B. CONTROLLED **BURNING** WITH COMBUSTION GAS CLEANUP METHOD

This is a complex area in which to pin down costs. The Army, Navy, and Air Force are all involved in various methods of controlled burning with combustion gas cleanup.

The Army at Picatinny Arsenal is working with pilot plant operation of induced draft, rotary kiln, and fluosolid incinerators. Their pilot plant capacity is in the **250** pounds per hour range. They plan conversion of an induced draft furnace to a 1,350 pounds per hour fluosolid incinerator.

The propellant (or explosive) washed out of the rocket motor is shredded under a water stream and transported as a slurry to the various incinerators. All but the fluosolid incinerator (which requires a cyclone separator to remove the particulate matter) require scrubbing of combustion gases. No capital or operating costs are available at this time.

**<sup>1</sup>**"Incheration Processes for Propellant and Explosive Waste Disposal;" Joseph S. Santos and John **J.** Canavan; Facilities and Protective Technology Division, Manufacturing Technology Directorate, Picatinny Arsenal

**<sup>1</sup>**The Navy at Indian Head, Maryland is in the process **d** developing a motor reclamation and propellant disval complex. **A** concept sketch of the complex is shown in Figure 1. The propellant washed out of a rocket motor is shredded with water and **the** resultant slurry fed to a wet oxidation reactor system which operates at 450" F and **600** psi.

The reported capital cost of the entire complex is \$20,000,000.

The high pressure wet oxidation unit, shown in Figure 1, which would have the capacity of 5 tons per day (417 pounds per hour), is estimated to cost \$800,000 to build. The labor and maintenance costs for the wet oxidation reactor are estimated at \$15 per ton or 0.75 cents per pound.

The Air Force at Edwards, California, has under construction a facility for incineration of solid and liquid propellants. **A** rotary kiln is used for the solid propellant and a special burner is used for the liquid oxidizers and fuels. All combustion gases are scrubbed. The design capacity is 500 pounds per hour of solid propellant and 600 pounds per hour of liquid propellant. The capital cost was reported as approximately \$3,000,000. The operating cost has not been determined, therefore, the cost per pound has not yet been calculated.

#### C. INGREDIENT RECOVERY METHOD

#### 1. Capital Costs

The cost of an Ammonium Perchlorate (AP) recovery plant is estimated to be \$250,000. This would be a portable plant (shown in Figure 2) with the capability of processing 21,000 pounds per day (or 875 pounds per hour) of composite propellant. It would produce 13,965 pounds per day of pure and damp ammonium perchlorate.

<sup>1</sup>Indian Head Memorandum Report 73-240; "Industrial Preparedness Measure: Propellant Disposal/Reclamation Facility Design." Naval Ordnance Station, Indian Head, Maryland



COST ESTIMATE FOR COMPLETE COMPLEX - \$20,000,000 COMPLETION ESTIMATE - 3 TO 5 YEARS

Figure 1. Navy Demilitarization Facility at Indian Head, Maryland



Figure 2. Waste Propellant Ingredient Recovery System Concept (Portable)

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For an additional \$60,000, **sn** air oxidation equipped trailer can be added to recover the aluminum powder from the residues left from the AP recovery plant. At the high propellant rate of 21,000 pounds per day, 3,360 pounds per day of high quality aluminum could be produced. As an alternate, a \$75,000 transesterificationequipped trailer could be used to recover both 3,360 pounds per day of high-quality aluminum and 2,940 pounds per day of impure polymer.

#### 2. **Operating** Costs

In calculating the operating costs of an ammonium perchlorate recovery plant, the following assumptions were made:

- **1.** 21,000 pounds of waste propellant are processed in 24 hours (7,000 pounds per shift).
- 2. The propellant will contain 70 percent ammonium perchlorate; 95 percent of the AP is recovered.
- **3.** 20 percent downtime will be experienced.
- 4. 13,965 pounds of ammonium perchlorate will be recovered each day (4,655 pounds per shift).
- **5. One** operator and one helper would be required to operate the plant.
- 6. Program support would require two manhours per day.
- **7.** Laboratory support would require five manhours per day.
- **8.** Utility costs will be \$20.00 per day.
- 9. Transportation costs of ammonium perchlorate in Econ-0-Bin containers to Henderson, Nevada (allowing for 10 percent moisture) are 1.635 cents per pound.

### 3. Operating Cost Calculations

#### a. Ammonium Perchlorate Recovery



The resale value at Henderson, Nevada will be at least 16 cents per pound.

Savings are 16 cents per pound minus 8.92 cents per pound = 7.08 cents per pound. The expected waste propellant and oxidizer from Space Shuttle **and** other pmgrams projected from 1976 to 1989 will amount to 7,962,600 pounds of propellant and 2,800,000 pounds of scrap ammonium perchlorate. This expected waste represents 8,095,129 pounds of recoverable ammonium perchlorate. At 7.08 cents per pound, this would amount to \$573,135 savings on recovered ammonium **per**chlorate alone during the 1976 to 1989 time period.

b. Aluminum Recovery

**In** calculating the operating cost of an aluminum recovery (air oxidation) or an aluminum and polymer recovery (transesterification) trailer addition, the following assumptions were made:

> 1. The aluminum and polymer recovery trailer will be conveniently located close to the AP recovery trailers to allow monitoring of the operation,

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addition of residue, **ae** well as aluminum powder and polymer removal by operators from the **AP** recovery trailers.

- **2.** Instruments will be installed that will allow monitoring from the AP recovery trailer control room.
- **3.** The daily residue after the AP is removed will contain 3,360 pounds of aluminum and 2,940 pounds (350 gallons) of impure polymer.
- **4.** Utilities and materials will cost \$30 per day.
- **5.** Aluminum powders and impure polymer will be loaded in used barrels available on plant.
- **6.** The value of aluminum powder is 40 cents per pound from the air oxidation process and 75 cents per pound from the transesterification process.
- 7. The impure polymer will have a value of 23.6 cents per gallon (or 2.83 cents per pound) as a fuel.

### 1. Air Oxidation Alum'num Recovery

Cost Labor effort covered by AP Recovery Plant **\$0**  Laboratory Support (1 manhour per day average) Program Support (1 manhour per day) 19.65 (Direct-Direct) 24.53 (Direct-Direct) Utilities per day 30.00 TOTAL \$74.18

The assumed costs for aluminum recovery using the air oxidation method

are:

At 3,360 **pounds** per **day** of aluminum powder, **the** cost of process would be **2.2 cents** per pound. At a value of **40 cents** per **pound,** the savings would be 37.8 cents per pound.

The quantity of aluminum involved between 1976 and 1989 would amount to 1, 274, 016 pounds for a 37.8 cents per pound savings. The savings on air oxidation **aluminum** would amount **to \$481.St8.** 

#### 2. Transesterification Aluminum and Polymer Recovery



At 3.360 pounds per day of aluminum powder. the **cost** of processing would be **1.28** cenis per pound.

At a value of **75** cents per **pound** for high **puri!y** aluminum powder. the **savings** would be **70.72** cents per pound.

The quantity of aluminum involved between 1976 and 1989 would be 1,274,016 pounds. .It **70.72** cents **per** pound, the total **savings** on high purity aluminum would be **\$900.984.** 

reclaimed aluminum **(even** though it meets all of the specifications for use in rocket motors) may still be unsalatle for that use. It may **be** necessary to sell it at a lower price **(40** cents per pound). **'Ihe** savings then would be **35.72** cents **per pound** or for **1.274.016** pounds **x 35.72** cents per pound the savings is **\$455,018**  during the **14** year Space Shuttle production period. Selling the **1.115,000 porlnds (133.761** gallons) **of impure** polymer as fuel oil at **23.5** cents per gallon would result in a savings of **\$31,567** in the **1976** to **1989** time period.

The total savings on AP aluminum and polymer could amount to \$1.5 millions over the **Space Shuttle** production period.

**Items** that **COUM** increase the savings **are:** 

- 1. Operation of the Recovery Plant with sister divleion labor and thus **reduce** labor **coats.**
- **2.** Negotiate an AP **selling** prtce higher **than** 16 cents per pound.
- **3. Find** a sale for the polymer that would **be more**  lucrative than eelllng it for **fuel.**
- **4.** Increase the amount of waste propellant available by washing **out recycled** motors. such **as** First **Stage** Minutem **an (45,000 pounds** per mot0 r) . Third **Stage** Minuteman motors **(7,300 pounds** per motor), and **even** by **washing cut motors for** the Army and **Navy.**

The AP, aluminum. and polymer available from one First Stage Minuteman mot **r** would be:



The potential savings from one First Stage Minuteman Motor (45,000 lb of propellant) would be **\$5,000.** 

**Some** facility effort would be necessary to prepare the washed out propellant for **feeding** to the reclamation aystem.

#### **D. SLURRIED EXPLOSIVES**

#### **1.** Capital **Costs**

Slurried explosives are made from shredded propellant or residues (propellant with the ammonium perchlorate removed) mixed with an oxidizer (sodium nitrate or ammonium nitrate). a sensitizer **(TNT, PETN** or **HAIX)** and a gelling agent.

The facility costs should include a propellant shredder, if one is not available. Assuming that one is not available. the cost for a facility (see **Figure 3)**  to shred, blend. and package slurried explosives at a **-50.360** pound per **day** rate is estimated to be \$175,000.

#### 2. Operating Costs

The following assumptions **were made:** 

- 1. The slurried explosive will contain 41.7 percent shredded propellant. 23.1 percent ammonium nitrate. 5.8 percent sodium nitrate. 14.6 percent PETN (sensitizer). and **0.8** percent gel agent.
- 2. The quantity of slurried explosives produced will be based on the availability of a maximum of **21,000** pounds of waste propellant per day.

The material costs per day will be:





**Figure 8, Slurried Explosivr Production System Concept (Portable)** 

#### Cost Per Day



**3. The** unit cost for slurried explosive is 21 cents per pound. The sale price is expected to be **40** cents per pound. This would result in a savings of **<sup>19</sup>** cents per pound or \$9,568 per day.

Items that can affect the savings are:

Labor

- **1. The** capacity calculated is for the maximum conceivable amount of propellant available. Realistic quantities would be much lower. The **daily capacity could be kept up and operating days** reduced, or the daily capacity reduced and more operating days utilized. The unit cost will increase if the daily capacity is reduced due to the additional labor required per pound.
- 2. The substitution of available scrap HMX **as** a sensitizer would result in a drastic reduction in raw materials cost. The sensitizer **(PETN)** cost amounts to **79** percent of the cost. If this could be done. the unit cost for slurried propellant would drop to **6.3** cents per pound and the savings would be 33.7 cents per pound or **\$16,934** per day.

#### **E.** DIRECT **USE** OF PROPELLANT AS FIRE STARTERS

It is expected **that** a limited quantity (10,000 **pounds** per year) of waste propellant would be disposed of in the iom of fire **starters.** The Forest Service **has** shown an interest in **these** starters. Some of these fire starters have **been**  fabricated and tested successfully inhouse. Twenty units have been fabricated for the Forest **Service** and it is anticipated that the Forest Service will test them this year. **The expected** selling price will be \$3.25 each.

The cost of producing the l-gallon fire starters **would** be:



**Thus** for 240 units per shift (at \$478.08 per shift) fabrication cost per unit is \$1.99; this would utilize 14 lb per unit or 3,360 lb of propellant per shift.

Items that affect the savings are:

- 1. If the operating personnel can place the fire starters in the shipping containers the cost of shipping will be reduced.
- **2. The** Forest Service is expected to pick up the fire starters at Thiokol and to save the shipping containers so they can be reused.

**3.** It is also possible that at least part of the labor presently **used** to **package** waste propellant in waste containers can be utilized to package the waste propellant in fire starters and thus reduce the cost of producing them.

## **III. ENERGY CONSERVATION EVALUATIONS OF WASTE PROPELLANT DISPOSAL METHODS**

#### **A. OPEN BURNING METHOD**

When waste composite propellant is burned, **1,800** Btu (or **0.52 KWH)** per **pound** of energy is liberated and lost. This does not include the energy **consumed** in making the ingredients or processing the propellant.

**<sup>A</sup>**slight amount of energy **(0.1 KWH** per **pound)** is also consumed in collection, placing, and igniting the propellant.

The reason that the Btu per pound yield for composite propellant is low. is that 70 percent of the propellant (by weight) is not fuel but oxidizer. Normal fuel uses oxygen from the air when it burns; but even with the energetic materials in propellant, pound for pound Bunker "C" fuel oil and propellant are about equal in energy yield.

#### **B. CONTROLLED** BURNING **WITH COMBUSTION** GAS **CLEANUP METHOD**

Rotary kilns, fluosolid reactors, and wet oxidation methods of propellant incineration are all designed without any heat recovery system; therefore, there is no net recovery of the energy liberated during burning.

However, in most of the processes some of the heat liberation is utilized to help maintain the combustion temperatures once it has **been** reached.

Auxiliary fuel (usually oil) is used to get the incineration equipment up to operating temperature. Electrical requirements for fans, rotating mechanisms.

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air compressors, scrubbers, pumps and control equipment are energy consuming. For these reasons the energy **used** or lost is estimated to be 0.9 **KWH** per pound for the controlled burning of waste propellant with **gas** scrubbing.

#### **C. INGREDIENT RECOVERY METHOD**

In the above methods all materials are destroyed. In the proposed ingredient recovery system 0.3 **KWH** per pound of propellant would be **used** to operate the recovery plant. The energy conserved by ingredient recovery, based on replacement energy requirements. would amount to 9.8 **KWH** per pound of propellant. 1. **1. Produce Propellant Ingredient** Produce Propellant Ingredients - 1. **Energy Requirements** to Produce Propellant Ingredients

#### a. Ammonium Perchlorate

Ammonium perchlorate is made by double electrolysis of sodium chloride salt solution.

In the first electrolytic step, sodium chloride is processed to yield sodium chlorate.

In the second electrolytic step. sodium chlorate is processed to yield sodium perchlorate. Sodium perchlorate is then reacted with ammonium chloride to produce ammonium perchlorate.

Recent data from an ammonium perchlorate producer shows 5.3 to 5.7 **KWH**  per pound of AP for the chlorate electrolysis and 1.8 to 1.9 **KWH** for the perchlorate electrolysis, or an average of 7.1 to 7.6 KWH per pound of AP produced. This contains an allowance for the boiler and refrigeration, but nothing is added for the production of hydrochloric acid and anhydrous ammonia. both of which are used to make the ammonium chloride, nor for the **drying** of the ammonium perchlorate. Including all of the above. it is estimated that at least 9 **KWH** of energy are required to make one pound of ammonium perchlorate.

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#### b. Aluminum Powder

Aluminum is produced by the electrolysis of refined alunite (aluminum oxide) dissolved in cryolite (aluminum fluoride at a temperature above **725"** C **(1,336' F).**  The energy required to produce one pound of aluminum metal from the alunite is 9.5 **KUTH.** By the time the energy cost of producing aiunite from bauxite clays, and the cost of producing an aluminum powder are added in. the energy requirements are near **14 KWH** per pound of aluminum powder produced.

c. Polymer and Epoxy Curing Agent (ECA)

The heat of formation of polymer is **10.800** calories per **gram.** This would **be 19.440** Btu per pound or 5.7 KWH per pound, which is a conservative figure. Allowing for processing equipment operation and efficiencies, a value of 9 KWH per pound would be more realistic. This number will be used for both polymer and ECA because 01 the similarity of their preparation.

#### 2. Energy Requirements for Proposed Ingredient Recovery Process

The following assumptions are made:

- **1.** The plant. when operating at full capacity. will process **21.000** pounds of propellant per day.
- 2. The propellant shredder will operate at  $75\%$ load for **10.5** hours per day.
- 3. The process water stream will average 5 gallons per minute and be heated to **190°F** (88'C).
- **4.** Crystallization will take place at 95°F (35°C).
- 5. **or heating pipes, tank jackets, and space** ting the heat requirement will be tripled.
- $6.$  The transesterification method of residue treatment will be used in the calculations.

- Energy Requirements Per **my** 



When processing **21,000 pounds** of propellant per day, the energy requirements per **pod** would be **0.31 KWH.** 



The energy value of the ingredients of one pound of propellant are:

Total Energy Required to Replace 1 Pound of Propellant **9.80** 

The net energy saving per pound of propellant processed would be **(9.8** minus **0.31) 9.49 KWH.** The Space Shuttle program is expected to produce **7,962,600** pounds of propellant and **2,800.000** pounds of waste ammonium perchlorate in the **14** year production period. The energy savings that could be expected would amount to **75.6** million KWH of power for the processed propellant and 24.3 million KWH for the waste ammonium perchlorate processed. This savings would amount to approximately **100** million **KWH** of energy in the **14** year period.

Figure **4** graphically illustrates the energy relationship between disposal methods and points out the energy advantage of ingredient recovery.



Figure 4. Energy Relationship for Various Disposal Methods

#### IV. - CONCLUSIONS AND RECOMMENDATIONS

#### **A.** CONCLUSIONS

The economics of waste propellaat disposal indicate that the shift from open burning to one of the controlled burning methods is costly. New equipment costs as well as higher operating  $\omega$ sts will be encountered.

The recovery of ingredients from waste propellant (when there is a large supply of composite propellant) can be a self-supporting operation. It is predicted that for the Space Shuttle program over \$1.5 millions can be saved by recovery of ammonium perchlorate and aluminum.

The energy study balance is also in favor of ingredient recovery. The savings in energy on the same Space Shuttle program would amount to **100** million KWH of energy, while open or controlled burning represent a total loss of energy.

#### **B. RECOMMENDATIONS**

It is recommended that the detailed design of a recovery plant be initiated.

It is also recommended that a plant funding effort be initiated. The goa! should be to have an ingredient recovery plant onstream prior to the start of the Space Shuttle production effort in 1977.

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