JPL PUBLICATION 80-54

NOU-33317

(NASA-C2-103504) LECHNIQUES FOR MEASURING ARBIVAL TIMES OF PULSAR SIGNALS 1: DSN UBSERVATIONS FROM 1968 TO 1980 (Jet Propulsion Lab.) Ed p HC AU5/MF AU1 Unclas CSCL 03A G3/89 28761

Techniques for Measuring Arrival Times of Fulsar Signals I: DSN Observations from 1968 to 1980

G.S. Downs P.E. Reichley

August 15, 1980

National Aeronautics and Space Administration

Jet Propulsion Laboratory California Institute of Technology Pasad ana, California



Techniques for Measuring Arrival Times of Pulsar Signals I: DSN Observations from 1968 to 1980

G.S. Downs P.E. Reichley

August 15, 1980

National Aeronautics and Space Administration

Jet Propulsion Laboratory California Institute of Technology Pasadena, California

ABSTRACT

Natural radio emissions from pulsating radio sources (pulsars) have been detected at Goldstone on a regular basis since 1968. Scientific analysis of these signals has stimulated ideas of traversing the interplanetary medium and beyond, to the "home" of comets, using these natural beacons as navigation aids. Therefore, the techniques used in the groundbased observations of pulsars are described here in the required detail, many of them being cirectly applicable in a navigation scheme. The arrival times of the pulses intercepting Earth are measured at time intervals varying from a few days to a few months. Low-noise, wide-band receivers, unique to the stations of NASA's Deep Space Network, amplify signals intercepted by the 26-m, and 34-m, and 64-m antennas at Goldstone and in Spain. Digital recordings of total received signal power versus time are cross correlated with the appropriate pulse template, thereby providing an estimate of the pulse arrival time relative to the station clock. Corrections are applied to the station clock to obtain arrival times relative to ephemeris time. Drifts in phase encountered during signal integration are removed. The arrival times are then referred to the barycenter of the solar system for scientific studies, and to the geocenter for export to other investigators.

PRECEDING PAGE DLARL NOT FRAME

iii

CONTENTS

distant de la constante

i. Manie

3

;

TT - the concounted

Ι.	INTRO	DUCTION	1
11.	DATA	COLLECTION	5
	Α.	RECEIVING, DETECTION AND RECORDING	5
	В.	THE SYSTEM CONFIGURATION CODES	8
	с.	EFFECTS OF RECEIVER PARAMETERS ON THE ARRIVAL TIME	17
111.	EDITI	NG AND COLLATING	21
	Α.	EDITING THE DATA	21
	В.	COLLATING THE DATA	23
IV.	PULSE	S-SHAPE TEMPLATES	24
	Α.	CORRELATION WITH A TRIANGLE	25
	Β.	CORRELATION WITH THE TRUE TEMPLATE	27
	с.	SUBTRACTION OF THE BACKGROUND	29
	D.	THE TEMPLATES	30
	E.	TEMPLATE SMOOTHINC	31
	F.	CENTRAL SPIKES	46
	G.	DEFINING THE ZERO LEVEL	46
۷.	ESTIM	ATING THE PULSE ARRIVAL TIME	48
	Α.	THE CORRELATION PROCESS	48
	В.	ESTIMATING THE DELAY	52
	С.	ADJUSTMENTS	55
	D.	UNCERTAINTY IN THE ARRIVAL TIME	55
vī.	GEOCE	NTRIC ARRIVAL TIMES	5 7
	Α.	CORRECTION FOR PULSE SMEARING	57
	в.	ANTENNA POSITION	63
	с.	DISPIRSION	67

PRECEDING PAGE DI ANN NOT

	D.	DOPPLER DISPERSION	67
	E.	ARRIVAL TIME IN COORDINATED UNITERSAL TIME	68
	F.	CONVERSION FROM COORDINATED UNIVERSAL TIME TO EPHEMERIS TIME	69
	G.	AN EQUIPMENT IDIOSYNCRASY	71
	н.	A TEST OF THE MEASUREMENT CONSISTENCY	71
VII.	THE T.	ABULAR RESULTS	73
	REFER	ENCFS	78

Figures

.*

1.	Nomenclature of the Pulsed Emission	7
2.	The Receiving, Detection, and Data Recording of Signals from Pulsars	9
3.	The 1-Second Pulses from the Station Clocks at DSS-13 and DSS-14, Goldstone, and Their Relation to UT (station)	12
4.	The Timing Control Box for Pulsar Data Collection	14
5.	Three Triangular Functions Overlying a Data Array	26
6.	The templates of PSR 0031-07: (a) P = 0.943 sec; (b) Expanded time scale	32
7.	The template of PSR 0329+54: (a) P = 0.715 sec; (b) Expanded time scale	32
8.	The template of PSR 0355+54: (a) P = 0.156 sec; The pulse has been smoothed by a window 160 µsec wide. (b) Expanded time scale	33
9.	The template of PSR 0525+21: (a) P = 3.745 sec; The pulse has been smoothed by a window 2.2 ms wide. (b) Expanded time scale	33
10.	The template of PSR 0628-28: (a) P = 1.244 sec; (b) Expanded time scale	34

....

~

Figures

..**د.**

- .-

· •

.

Ϊ,

•

11.	The template of PSR 0736-40: (a) P = 0.374 sec; (b) Expanded time scale	34
12.	The template of PSR 0823+26: (a) P = 0.531 sec; (b) Expanded time scale	35
13.	The template of PSR 0833-45: (a) P = 0.089 sec; (b) Expanded time scale	35
14.	The template of PSR 0950+08: (a) P = 0.253 sec; (b) Expanded time scale	36
15.	The template of PSR 1133+16: (a) P = 1.188 sec; (b) Expanded time scale	36
16.	The template of PSR 1237+25: (a) P = 1.382 sec; (b) Expanded time scale	37
17.	The template of PSR 1604-00: (a) P = 0.422 sec; (b) Expanded time scale	37
18.	The templates of PSR 1642-03: (a) $P = 0.388$ sec; The data has been smoothed by a window 230 µsec wide. (b) Expanded time scale	38
19.	The template of PSR 1706-16: (a) P = 0.653 sec; The data has been smoothed by a window 390 µsec wide. (b) Expanded time scale	38
20.	The template of PSR 1749-28: (a) P = 0.563 sec; (b) Expanded time scale	39
21.	The template of PSR 1818-04: (a) P = 0.598 sec; (b) Expanded time scale	39
22.	The template of PSR 1911-04: (a) P = 0.826 sec; (b) Expanded time scale	40
23.	The template of PSR 1929+10: (a) P = 0.227 sec; (b) Expanded time scale	40
24.	The template of PSR 1933+16: (a) P = 0.359 sec; (b) Expanded time scale	41
25.	The template of PSR 2016+28: (a) P = 0.559 sec; (b) Expanded time scale	41

-

...

. .

.

Figures

.

27.	The template of PSR 2045-16: (a) P = 1.962 sec; The data has been smoothed by a window 1.2 ms wide, (b) Expanded time scale	42
28.	The template of PSR 2111+46: (a) P = 1.015 sec; (b) Expanded time scale, with the data smoothed for display by a window 2.0 ms wide	43
29.	The template of PSR 2217+47: (a) P = 0.538 sec; (b) Expanded time scale, with the data smoothed for display by a window 1 ms wide	43
30.	Shifting and Transforming the Template	49
31.	Computing the Cross Correlation Coefficients	51
32.	An Example of the Cross Correlation Process; PSF 1818-04 on 27 Jan. 1970 at 16h 58m 00 ^s UT	52
33.	Correlation Coefficients Versus Time Delay Near the Peak Shown in Figure 32	55
34.	Geometry for Calculating the Phase Drift	59
35.	Nomenclature for Enumerating Pulses	60
36.	An Example of Drift in the Phase of PSR 0833-45 on 1 Dec. 1968 at 11 ^h 00 ^m 00 ^s (UT)	63
37.	Time-of-Arrival Residuals of Measurements Made at DSS-13 and DSS-14 on the Modified Julian Date (MJD) of Observation for PSR 0833-45	72
Tables		
1.	24 Pulsars Monitored at JPL at Frequencies of 2295 MHz and 2388 MHz	3
2.	Chronology of the Major Experiment Configurations for Collecting Pulsar Timing Data	10
}.	The Configuration Codes Corresponding to Non-Standard Observing Frequencies	11
4.	Pulse Smearing Due to Dispersion and Post-Detection Filtering	19
5.	Geocentric Antenna Coordinates	64
6.	Relations Used in Computing the GHA of an Observing Station	64

Ţ

viii

×

Tables

.

7.	Elements of th	he Precession Matrix	66
8.	PSR 0823+26:	Geocentric Times-of-Arrival	74

SECTION I

INTRODUCTION

Natural sources of pulsating radiation were first observed in 1967 (Refs. 1, 2) by radio astronomers at Cambridge University in England. These pulsating radio sources (pulsars) are exciting because they (1) are thought to represent a stage in stellar evolution known as neutron stars (Ref. 3), providing a probe of an extremely dense state of matter; (2) exhibit changing, often dramatically so, behavior in the pulse period, signifying changes in the spin rate of the neutron star driving the pulse generating mechanism; and (3) have large space velocities, implying violent origins.

A series of measurements is being established using the NASA-JPL Deep Space Network (DSN) which will allow a probing of the neutron star interior and a direct measurement of changes in spin rate and angular motion. The purpose of the measurements is to measure the phase of the pulse train at known epochs. It requires little imagination to realize that the simultaneous comparison of the phase of a pulse train at two different DSN stations represents either a measurement of the difference between the station clocks, or a measurement of the loca+ion of one station relative to the other. In navigation, one assumes the clocks are synchronized. Differences in pulse phase are then attributed to differences in location. We need only to replace one DSN station with a spacecraft to complete the illustration.

The purpose of this report is to describe in appropriate detail the techniques used currently in the DSN for measuring the arrival time

of a particular pulse (alternatively, the phase of the pulse train relative to a given epoch). Phase measurements were begun in September 1968 and continue to the present. The 24 pulsars included in this series are listed in Table 1 with appropriate dates on which measurements began. An example of the results of this long series of measurements is presented in Table 7.

Many characteristics of the pulsars have been or can be deduced from the measurement of pulse phase at a series of epochs if the epochs occur often enough (weekly to monthly) and over a sufficient time span (years). Usually a parameterized model of the pulsar is constructed which is used to predict the time-of-arrival of a given pulse (or the phase of the pulse train at a given epoch). The parameters are then adjusted to minimize the mean square difference between the predictions and the meas oments. The final values of the model parameters are dependent on the planetary ephemeris used in the data reduction, since all observations are usually referred to the barycenter of the solar system. The importance of this is clear when one realizes that 2 milliseconds flight time in the displacement of the barycenter amounts to as much as 1 arc-second displacement in the position of the pulsar. Errors in the knowledge of the barycenter with time scales of tens of yoars are caused by inaccuracies in the values of the masses of the outer planets. These slow curvilinear errors affect the values of the higherorder derivatives of the pulse period. Hence, though all current ephemerides may produce similar pulsar model parameters, improved ephemerides available twenty years from now will produce significantly different results. Tables of geocentric arrival times, being important

Entry	Pulsar	Sta	art I	Date
1	0031-07	5	Mar	1970
2	0329+54	5	Sep	1968
3	0355+54	5	May	1973
4	0525+21	1	Jun	1969
5	0628-28	30	Dec	1969
6	0736-40	1	0ct	1970
7	0823+26	12	Feb	19 69
8	0833-45	22	Nov	1968
9	0950+08	5	Sep	1968
10	1133+16	25	0ct	1968
11	1237+25	10	Aug	1969
12	1604-00	1	0ct	1970
13	1642-03	12	Jul	1969
14	1706-16	1	0ct	1970
15	1749-28	2	May	1969
16	1818-04	29	Dec	1969
17	1911-04	5	Mar	1970
18	1929+10	29	Jun	1969
19	1933+16	23	Dec	1968
.'0	2016+28	5	Sep	1968
24	2021+51	30	Dec	1969
, , , , , , , , , , , , , , , , , , , ,	2045-16	10	Dec	1968
23	2111+40	30	Dec	1969
24	2217+47	30	Dec	1969

Table 1. 24 Pulsars Monitored at JPL at Frequencies of 2295 MHz and 2388 MHz

.

~

, ***** .

y something a statement as

ì

۰**۰**۰

3

الم وحمو د

.

archives of pulsar behavior, are independent of small changes in the ephemerides of the planets. They form the exported form of the observational results.

The measurements were performed earlier at a frequency of 2388 MHz on the 26-m antenna at Deep Space Station 13 (DSS 13) and the 64-m antenna at DSS 14. Current observations are at 2295 MHz, utilizing all facilities at Goldstone and in Spain. Detected pulsar signals are sampled, integrated and recorded on magnetic tape. Data collection and recording is discussed in Section II. The data are then edited and, for the early years, collated according to the pulsar, as described in Section III. The collated data, stored on magnetic tapes, was used to derive a template for each pulsar, representing the average power level across a pulse of radiation. The template is then cross correlated with each data record of the particular pulsar to obtain an estimate of the pulse arrival time. The procedures used in constructing the templates is described in Section IV and the cross correlation process is discussed in Section V. The correction of the arrival time estimates for various small effects is discussed in Section VI. A sample list of geocentric times-of-arrival is presented in Section VII.

SECTION II

DATA COLLECTION

A. RECEIVING, DETECTION, AND RECORDING

「御寺町で

14

)) It was shown shortly after the discovery of pulsars that they could be detected at 2295 MHz (Ref. 4). Shortly thereafter, in August 1968, JPL observations at 2388 MHz were b⁻;un using the 26-m antenna of DSS 13 at the Goldstone complex. The wide bandwidths (about 12 MHz) and the low system temperatures (about 16 K) allowed detection of three of the first four pulsars discovered. Over the following months the observations have been expanded to include 24 objects, utilizing all antennas at Goldstone and Madrid.

The receiver consists of the antenna, a right circularly polarized feed horn (left circularly polarized at 2388 MHz), and a maser preamplifier followed by conversion to and amplification at intermediate frequencies (IF). Low-loss cabling then carries the signal to the receiver room for more IF amplification. The receiver bandwidth is limited to 12 MHz by either the IF portion of the receiver or by a filter inserted for that purpose. Square-law detection of the signal is then performed, followed by low-pass filtering. The output of the low-pass filter is amplified by a wideband DC amplifier to a 1- to 2-volt level. Subsequently, the signal is converted to digital form in a sampling process. The integration time (see Ref. 5 for a definition) of the post-detection filter is chosen to match the time interval between analog-to-digital (A/D) conversions.

2 Yest at which as

The data processing, from sampling to the dumping onto magnetic tape, is under computer control. The detected signal is sampled at a

rate equal to 5000 times the apparent pulse period, corresponding to sample intervals of 50 µsec up to 800 µsec. Converted to a digital signal by an A/D converter, the samples are stored in the memory of an SDS 930 computer or a standard station Modcomp computer. Data samples are superimposed modulo the pulse period. Samples from 500 pulse periods are superimposed congruently. Then a 5000-word array representing the superposition of 500 pulses is dumped onto tape. This accumulation of pulses continues until (a) detection is ensured, or (b) it is clear the pulses are too weak at this time to be observed. Accumulations rarely run more than one hour.

The form of the recorded data is sketched in Figure 1. The pulse is sitting on top of a system noise power corresponding to a temperature T_s . The peak equivalent temperature of the pulse is T_p . The 5000 samples spanning one pulse period are plotted from left to right. The time of the first data sample, plotted at the left, corresponds to a particular epoch of Universal Time (UT) as recorded at the station. This start time is recorded in UT (station). The train of received pulses then has a particular phase relative to the start time. The pulse train is folded in upon itself to beat down the effects of receiver noise fluctuations. Rather than think in terms of the phase of the pulse train, we ask for the arrival time of the first pulse after the start time. Hence, the pulse delay is added to the UT (station) epoch corresponding to the start time to yield an epoch for the arrival time of the pulse.

The phase of the pulse train may not remain constant during the supe -position of many pulses. The subsequent drift in phase requires that a correction be applied to the measured pulse delay. This

6

a star a start a start a same a se a same



correction is discussed in Section VI, D. Small corrections are required in the start time and in the UT (station) defined by a particular clock. These corrections are discussed in this section and in Section VI.

2

B. THE STATEM CONFIGURATION CODES

Mart 11 . The Street of the Content

The measurement system is shown diagrammatically in Figure 2. The experimental hardware is grouped into units defined by the dashed boxes. Each subsystem so defined has been interchanged with other similar subsystems, causing significant changes in the measured phases. Hence, a system configuration code has been constructed to alert the analyst to changes in the system. The code varies from 0 to 21, representing 22 significantly distinct combinations of antenna, receiver, observing frequency, computer, data collection controller, and station clock. A summary of the basic configurations (0-5) and (17-21) appears in Table 2. Additional configuration codes (6-16) were created to denote the use of an observing frequency of 2295 MHz or 2328 MHz, and their relationship to the basic codes are shown in Table 3. Left circular polarization (LCP) is to be assumed unless specified as right circular polarization (RCP).

Since the system configurations do not change in principle from one to the other, only configuration 0 is described in detail. Highlights of the other configurations are noted in the following paragraphs. Configuration 0 includes the 26-m antenna at DSS 13. The radio frequency bandpass is determined by the IF amplifiers. The final IF amplification, detection, and low-pass filtering is done by a receiver (labeled as receiver 1) built in 1968 by C. F. Foster. The timing control box was designed by G. A. Morris and is labeled timer 1.



Chronology of the Major Experiment Configurations for Collecting Pulsar Timing Data Table 2.

- 10

141.00

۲ ۲

r

onfiguration Code	Station	Dates In Effect (d/m/yr)	Description
0 (early)	DSS 13	n5/09/68-10/01/74	26-m Antenna; Keceiver 1; Timer 1; 1-sec Tick Derived From Ce Frequency Standard; Slow A/D Converter
0 (late)	DSS 13	10/01/74-01/04/76	Changed From Receiver 1 to 2
4	DSS 13	01/04/76-Present	Changed from Timer 1 to 3
 `	DSS 14	05/09/68-14/04/73	Receiver 1 Plus 12-MHz IF Filter; Fast A/D Converter
2	DSS 14	14/04/73-01/11/74	Changed From Timer 1 to 2
۳	DSS 14	01/11/74-01/04/76	l-sec Pulse Widened From 100 μsec to 200 μsec; -0.996550 Seconds Added to Start Epoch if Juliar Date > 2442460, 5
Ś	DSS 14	01/04/76-Present	Timer 2 Replaced With Timer 1
17	DSS 62	01/04/79-Present	2295 MHz, RCP, using station TPA with Modcomp
18	DSS 11	01/12/79 + Present	2295 MHz, RCP, using station TPA with Modcomp
19	DSS 63	01/11/79-Present	2295 MHz, RCP, using station TPA with Modcomp
20	DSS 12	01/04/80-Present	2295 MHz, RCP, using station TPA with Modcomp
21	DSS 14	01/04/80-Present	2295 MHz, RCP, using station TPA with Modcomp

.

·····

· · · · · · · · ·

رايت در کار کارد وارد.

· * '*' *'

والمواقع والمراجع والمحالية والمتراكبين المحاربة والمحمج المحاربات المحاربين والمحملي والمحمل والمراجع المراجع

· . .

0 }* *

en and the set of the second second

- -

	· · · · · · · · · · · · · · · · · · ·	Observi	ng		
Configuration	Configuration	Frequency	(MHz)	Description	
8	0	2328			
9 ∮		2295			
10	1, 2, 3	2328		-	
11)		2295	•		
8	4	2328			-
7 🖇		2295			
6	5	2295			
12	Any			Denotes a correction w s made standard post-detection Δf_{ℓ}	to
13	4	2295		RCP	
14	5	2295		RCP	
15	5	2295		DSS 13 via link to DSS 14	
16	5	2295		DSS 12 via link to DSS 14	

Table 3. The Configuration Codes Correspondingto Non-Standard Observing Frequencies

ind trave

The relationship between the epoch-defining 1-second pulse and Universal Time as disseminated at the station, UT (station), is shown in Figure 3 along with the relationships for the other codes. Exact relations between UT (station) and Coordinated Universal Time (UTC) is discussed in Section VI. Data sampling in configuration codes less than 17 is performed by a 12-bit A/D converter with a 40 µsec conversion time. The samples are stored in an SDS 930 computer having a 16,384-word memory. The station telemetry processor assembly (TPA) is used in codes 17 or larger.

a a state with the state of the

ŝ



The final IF amplification and detection is performed in the steps shown in Figure 2. The first unit built was transported between DSS 13 and DSS 14 until 10 Jan. 1974, when it was permanently installed at DSS 14. This unit was replaced at DSS 13 by a programmable unit built by C. F. Foster (described in Ref. 6 and labeled receiver 2). This change is not noted by a configuration code change. Hence, code 0 consists of an early phase (pre-10 Jan. 1974, Julian date 2442057.5) and a late phase (post-10 Jan. 1974).

The timing control box used here is not described elsewhere, so a block diagram is presented here for tutorial purposes. Other control boxes used in other configurations operate on the same principle, but differ in complexity and detail. Careful study of Figure 4 shows that a string of fast and slow pulses are generated by the timing control box to control the data sampling. The pulses are derived by counting down pulses from a frequency synthesizer, starting at a well-defined time by gating the synthesizer output. The gate must be armed and then a 1-sec pulse must be received before counting begins. To assist the operator in determining on which second the counting started, the 1-sec pulses are counted down by 10 so that counting will start modulo 10 seconds.

The fast pulses, occurring every $P_a/5000$ seconds, where P_a is the apparent pulse period, are used to strobe the A/D converter. The slower pulses, occurring every P_a seconds, are used to control the data transfer into core memory. On the first slow pulse, a computer interrupt is coincident with the slow pulse commands the A/D converter to begin operation. About 20 µsec later the end-of-convertion (EOC) interrupt forces the computer to accept the new word, and about 20 µsec after

الم المالي - المُنْظَارُ المُنْسِينَ المُنْظِينَةِ فَقَامَ مُعَالَيْهِ اللَّهِ عَلَيْهِ اللَّهُ عَلَي الم



14

and the second second

: 18. -

interruption, the word has been added to the proper memory location. Note that although it takes time to set up the computer to accept data, the EOC pulse arrives after the setup, so the first sample accepted is, in fact, coincident in time with the 1-sec pulse modulo one pulse period. However, a configuration exists in which the first sample is <u>not</u> coincident, and a correction must be applied.

The data sampling and superposition is under the control of a program written by G. A. Morris. The program also controlled the writing of the results onto magnetic tape for further reduction.

The apparent period P_a is the result of the Doppler effect caused by Earth's motion about the barycenter of the solar system. Tables of P_a are computed using a barycentric ephemeris of Earth and a crude but adequate model of the pulsar. The model includes the celestial coordinates of the object as well as the barycentric model of the pulse period (usually a period P_0 at some epoch t_0 , and the first derivative \dot{P}_0). The values of P_a are tabulated at 15-minute intervals between rise and set for any given day. The operator then chooses the value corresponding to the middle of the intended interval of pulse integration.

Configuration 1 consists of the 64-m antenna at DSS 14, the same IF amplification and detection unit, and the same timing box used in the early configuration 0. A 12-MHz IF filter is added to the chain to limit the IF bandwidth. The computer is again the SDS 930. The A/D converter, however, responds within 5 µsec of being strobed with a convert command. The computer, requiring about 20 µsec to arm the interrupt and start the data flow, misses the first data sample. The data stream is then shifted $P_a/5000$ seconds later in time. This shift is as large as 0.8 msec and is added to the UT (station) recorded by the operator.

FLIND WERE HILLING

Timer 1 was designed to accept the rising edge of a positive 1-sec pulse as the definition of UT (station). The 1-sec pulse distributed at DSS 14 is negative-going, the rising edge being coincident with UT (station). The necessary pulse inversion then causes the timing box to begin counting earlier than intended by a time equal to the width of the 1-sec pulse. This amounts to 100 μ sec in configuration 1 and 200 μ sec in configurations 3 and 5.

A programmable timing box, designed, built, and programmed by A. G. Slekys (Ref. 7) replaced the original timing box at DSS 14. This timer is labeled timer 2, and pertains to codes 2 and 3. A failure occurred in timer 2 which caused the start times of code 3 recorded between Julian dates 2442460.5 and 2443105.5 (17 Feb. 1975 and 23 Nov. 1976) to be earlier than intended. The correction is a subtraction of nearly 1 second. The exact correction was determined from a preliminary fit of a model of the pulsar emission times to the data. A correction of -0.996550 seconds to the start time aligned the affected arrival times with the times measured at DSS 13 (code 0 and 4) and with those unaffected from DSS 14.

Timer 1 was replaced at DSS 13 on 1 April 1976 by a programmable data collector/timer designed by S. S. Brokl (Ref. 8) and programmed by K. I. Moyd. This timer is labeled timer 3, and pertains to configuration code 4.

Occasionally a non-standard value is used for Λf_{ℓ} in the postdetection filtering. If a correction is applied, the configuration code is 12, corresponding to a correction to code 0 for dates prior to 1970 and to code 5 after 1976. (None occur in the intervening years.)

a too to the second

During the first half of 1979, the SDS 930 computer was phased out of service and removed in June 1979. This created the need to transmit IF signals over the microwave communication links to DSS 14 for data recording. The required 12 MHz bandwidth was obtained from the 6 MHz bandwidth of the link by folding the IF spectrum about the midpoint during base-banding. This technique was also used in obtaining signals from DSS 12. The configuration codes 15 and 16, then, indicate the recording at DSS 14 of signals from DSS 13 and DSS 12, respectively.

The most recent configuration for collecting pulsar data requires the use of the telemetry processor assembly (TPA) found in all DSN stations except DSS 13. The scheme was constructed by J. and M. Urech at DSS 62. Many of the hardware functions of Figure 2 are duplicated by their software. The data output is still in the form of 5000-word records on magnetic tape. The configuration cod of this scheme for each station is listed in Table 2.

C. EFFECTS OF RECEIVER PARAMETERS ON THE ARRIVAL TIME

Pulsar signals are dispersive since the interstellar plasma introduces a frequency-dependent group velocity. A broadband pulse, covering the radio spectrum from 30 MHz to 10 GHz, will arrive at the receiving antenna and sweep through the receiver passband at a rate given by

$$\frac{B}{\Delta t} = KF^3/DM \ (MHz/sec) \tag{1}$$

where the bandwidth B is in MHz, the observing frequency F is in MHz, Δt is in seconds, DM is the dispersion measure in parsecs (pc) cm⁻³, and K = 1.205043 x 10⁻⁴. Let Eq. (1) define the amount of smear Δt in the received pulse shape when all the other parameters are known. At F = 2388 MHz, a dispersion measure of 10 pc cm⁻³ causes the pulse to

smear 75 μ sec when the receiver passband is 12-MHz wide. The smear is 750 μ sec when the dispersion measure is 100.

Computed values of smear are presented in Table 4. As long as the bandpass parameters remain fixed over the years, the dispersive effect should not affect the measured arrival times. However, changes in a filter element or the slight alteration in the tuning of the maser amplifier will change the <u>frequency</u> centroid of the passband. This, in turn, will change the <u>time</u> centroid of the detected pulse. The magnitude of this effect must be determined experimentally and has not been done. The effect is expected to be small since it is not the limits of the passband that change, but only the shape.

The response of the post-detection filter is chosen such that the integration time τ is close to the time interval $P_a/5000$. In the case of these simple RC filters, $\tau = 2$ RC (Ref. 5), where R and C are the filter resistance and capacitance, respectively. Considerable care is exercised in using the appropriate filter combination for each pulsar throughout the observations. Use of non-standard values of τ could cause up to 200 µsec of time shift in the arrival time. The values are listed in Table 4. Note that the change from pre-10 Jan. 1974 (Code 0) to post-10 Jan. 1974 (Code 0) is principally a change in the post-detection filters, and much care was taken to duplicate the previous values of τ in the new filters.

A quantitative measure of the system performance is given by the ratio of σ , the root-mean-square (rms) value of the noise fluctuations about the mean, divided by ζ , the mean. Letting N represent the number of pulses superimposed,

18

 $\frac{\sigma}{\zeta} = \frac{1}{\sqrt{NTB}}$

1.

Pulsar	Dispersion Measure* (pc cm ⁻³)	Smear for B = 12 MHz (µsec)	Integration Time (µsec)
PSR 0031-07	10.89	79	200
0 329+5 4	26.776	195	50
0355+54	57.03	416	50
0525+21	50.955	372	400
0628-28	34.36	251	200
0736-40	161.0	1174	100
0823+26	19.4634	142	400
0833-45	69.08	504	50
0950+08	2.969	22	50
1133+16	4.8479	35	50
1237 +25	9.296	68	200
1604-00	10.72	78	100
1642-03	35.71	261	100
1706-16	24.88	182	1.50
1749-28	50.88	371	400
1818-04	84.38	616	150
1911-04	89.43	653	150
1929+10	3.176	23	50
1933+16	158.53	1157	50
2016+28	14.176	103	50
2021+51	22.580	165	100
2045-16	11.51	84	200
2111+46	141.50	1033	200
2217+47	43.54	318	100

如何的钟,李 恭王,你,"

1.444

4

1. 14.3 -

Table 4. Pulse Smearing Due to Dispersion and Post-Detection Filtering

*Values From Ref. 22

. ***

Ş

One expects for N = 500, τ = 50 µsec and B = 12 MHz, that $\sigma/\zeta \sim 0.0018$. Throughout the years of data collection this parameter has varied between 0.0013 and 0.0025, corresponding to 42 µsec < τ < 59 µsec. Hence, the values of integration time listed in Table 2 were stable to within 16 percent of the tabulated value. of South Landson

SECTION III

EDITING AND COLLATING

A. EDITING THE DATA

Successfully measuring the time-of-arrival of a pulse near a given epoch does not ensure that the result is free of biases. The editing process involves detecting the biases, correcting the measurement if the reason for the bias is known, or omitting the measurement if the reason for the bias is known but does not allow correction. A good example of a correctable situation is the use of a receiver frequency other than the 2388 MHz frequency used in most of the measurements.

A preliminary model (position and pulse period) is devised for each pulsar. These models are used in a computer program of adequate accuracy to predict pulse arrival times at the barycenter of the solar system. A barycentric ephemeris of the Earth is used to refer the measured arrival times to the barycenter and a residual (measured minus predicted) arrival time is computed. The occurrence of a residual that is large compared to the average fluctuation is cause for further investigation.

Several properties of the pulsar or the system are directly responsible for the need to correct or omit the arrival time measurement. They are:

(1) Scintillations

Interstellar clouds of ionized hydrogen impose variations on the received pulse power. The time scale of the fluctuations is on the order of tens of minutes. If the pulsar is far enough away, and several are, large fluctuations in the pulse power force the signal down into the receiver

noise for many minutes, causing a spurious estimate of the pulse arrival time.

(2) Start Times

The early timing box divides the incoming 1-sec pulse by 10 (see Fig. 4), allowing the counting to be started at well-defined 10-sec intervals. It is imperative that the counter status register be synchronized with the station clock, modulo 10 seconds; i.e., on the even 10-second mark, the counter should read 0. Occasionally synchronization is lost. In these cases it is easy to deduce how many integral seconds (never more than 10) the start time has to be retarded or advanced to obtain the true start time.

(3) Receiver Frequency

Dispersion caused by the interstellar plasma causes the arrival time to be a function of observing frequency. Using a non-standard receiver frequency produces a correctable error (see Section VI, F).

(4) Post-Detection Filter

Occasionally a non-standard filter width was used, resulting in unusual time delay through the filter. A correction is readily applied.

During the summer months of 1969 a bad detector connection caused unusually large (~1 ms) time delays. The corrections for this effect were derived by computing the autocorrelation function of the system noise, the width of which then gave the effective time constant of the filter. Both situations are labeled as system code 12 (see Section II, A).

(5) Miscellaneous

In a very few cases the reasons for & large residual are unknown. Common experimental situations in such cases are receiver tuning problems, repeated power failures, and radio frequency interference (RFI). The measurement results are usually omitted.

B. COLLATING THE DATA

A data tape produced at Goldstone contains data files corresponding to the pulsars observed during a particular observing session. This format is inconvenient for producing average pulse profiles. (These profiles, discussed in Section IV, are a necessary part of optimally estimating the pulse arrival time). Therefore, the early data files of a particular pulsar were gathered onto one or more magnetic tapes in the order they were collected at Goldstone. Preliminary editing took place during this process, to delete obviously faulty data. The main cause for omission at this stage is RFI. These data were used to compute the mean pulse profile presented in Section IV.

Data collected after August of 1972 is not collated according to pulsar. Data files are merely written onto magnetic tapes in the order they were collected without regard to the particular pulsar. The files were condensed by editing the data and filling each tape completely. Estimates of arrival times are obtained after each tape is filled (every 2 to 3 months).

SECTION IV

PULSE-SHAPE TEMPLATES

In the presence of white noise the maximum signal-to-noise (S/N) ratio for the output of a particular filter can be obtained when thc filter response has the same form as the input signal (a matched filter). The required matched filter is realized by cross correlating a received pulse with an estimate of the noise-free shape of the pulse. The estimates of the noise-free pulse shapes are obtained by superimposing many thousands of pulses to average down the effects of noise. The result is referred to herein as a template.

The proper registration or alignment of the pulses needs to be obtained to build a template with a minimum of smearing. Two passes, each described below, are used to optimize the registration. Correlation with a triangular function is first performed to obtain first-order estimates of pulse position within the data records. The template position in the data array corresponding to the maximum cross correlation coefficient is defined to be the pulse position. The numbers in each data record are then shifted circularly by the amount required for proper registration with all the other data records. Adding all the records produces a preliminary template. This template is cross-correlated with the original data to provide new estimates of pulse positions within the data arrays. The second result of the addition of all the properly registered data is taken as the correct average pulse shape. F ally, the central 4096 words of the 5000 word array are used to form the template. See Section V regarding the choice of the 4096 words.

A. CORRELATION WITH A TRIANGLE

In the early analyses of pulsar data (Ref. 9), the data were cross correlated with a triangular function closely matched in width to the pulsar signal. The pulses resemble the triangular function so the correlation function provides an adequate estimate of the pulse position within the data array.

The shape of the triangular function lends itself to a particularly fast algorithm for computing the cross correlation coefficients. Let the data record be presented by f_n , $1 \le n \le 5000$. Then correlate with the function

$$T_{r}(n, n_{0}, W) = \begin{cases} 1 - \frac{|n-n_{0}|}{W}; \frac{|n-n_{0}|}{W} \leq 1\\ 0 ; \text{ otherwise} \end{cases}$$
(2)

W is the half-width of the base of the triangle and n_0 is the position of the peak in the data array. This function is shown in Fig. 5 for $n_0 = W$, W+1, and W+2. Inspecting Fig. 5, the correlation coefficients C_0 and C_1 can be written as

$$C_0 = \sum_{n=1}^{2N-1} T_r(n, W, W) f_n = f_W + \sum_{n=1}^{W-1} \frac{n}{w} f_n + \sum_{n=W+1}^{2W-1} \frac{2W-n}{W} f_n$$

and

$$C_{1} = \sum_{n=2}^{2N} T_{r}(n, W+1, W) f_{n} = f_{W+1} + \sum_{n=2}^{W} \frac{n-1}{W} f_{n} + \sum_{n=W+2}^{2W} \frac{2W+1-n}{W} f_{n}$$

The difference between the coefficients becomes

$$C_1 - C_0 = -\frac{1}{W} \sum_{n+1}^{W} (f_n - f_{W+n})$$
 (3)



Figure 5. Three Triangular Functions Overlying a Data Array

This is the first quantity computed.

In a similar manner the difference between coefficients $\rm C_2$ and $\rm C_1$ is

$$C_2 - C_1 = -\frac{1}{W} \sum_{n=2}^{W+1} (f_n - f_{W+n})$$
 (4)

This is the second quantity computed, but in the following manner. This new quantity is obtained from Eq. (3) by adding the term

$$\frac{f_1}{W} - \frac{2f_{W+1}}{W} - \frac{f_{2W+1}}{W}$$

In general the new difference in adjacent coefficients is computed from the previous difference by

$$C_{\ell} - C_{\ell-1} = C_{\ell-1} - C_{\ell-2} + \frac{1}{W} (f_{\ell-1} - 2f_{W+\ell-1} + f_{2W+\ell-1})$$
(5)

26

j,

The value of C_{ℓ} relative to the coefficient C_{1} is calculated by accumulating the sums $C_{\ell} - C_{\ell-1} + C_{\ell-1} - C_{\ell-2}$ from Eq. (4) through the many steps suggested by Eq. (5):

$$c_{3} - c_{2} + c_{2} - c_{1} = c_{3} - c_{1}$$

$$c_{4} - c_{3} + c_{3} - c_{1} = c_{4} - c_{1}$$

$$\vdots$$

$$c_{\ell} - c_{\ell-1} + c_{\ell-1} - c_{1} = c_{\ell} - c_{1}$$

The position of the peak of the triangular function corresponding to the largest value of $C_{l} - C_{l}$ denotes the position of the pulse in the data array.

B. CORRELATION WITH THE TRUE TEMPLATE

an an the stranger and a set as the set of the

The cross correlation function $C(\tau)$ of two continuous functions is written as

$$c(\tau) = \int p(t)e(t - \tau) dt$$
 (6)

where p(t) is the pulse power as a function of time t, e(t) is the expected pulse shape (the pulse template) and τ is the displacement in time of the two functions relative to some well-defined reference time. In the discrete case encountered here, the correlation coefficients can be evaluated by directly performing the operation suggested in Eq. (6). However, it is much faster to use Fourier analysis techniques. Direct evaluation of Eq. (6) requires 2.5 x 10⁷ multiplications, while using the fast Fourier transform (FFT) algorithm requires only about 1.4 x 10⁵ multiplications.

Consider the Fourier transforms of the functions involved. The transform of the correlation function $c(\tau)$ becomes

$$C(s) = \iint p(t) e(t - \tau) dt e^{-i2\pi s\tau} d\tau$$

where s is the transform variable. Interchanging the order of integration,

$$C(s) = \int p(t) e^{-i2\pi s\tau} \int e(u) e^{i2\pi su} du d\tau$$

or

$$C(s) = P(s) E(-s)$$
⁽⁷⁾

where P(s) and E(s) are the transforms of p(t) and e(t), respectively. Hence, the procedure is to calculate E(-s) for each template (calculated only once), P(s) for the data record, and then the complex function C(s). Inverse transformation of C(s) yields $c(\tau)$.

The FFT algorithm is used in a form that takes advantage of the fact that a Fourier transform of a real function is the sum of a real even function and an imaginary odd function. Hence, if the data array is L samples long, only L/2 values need be calculated in the transform. See Ref. 10 for the original work on this form of the algorithm or the detailed discussion of its use by Brigham (Ref. 11). Briefly, the data is placed in an array L samples long, where L is a multiple of 2. Pretending that the array is complex of dimension L/2, an FFT is performed on the complex array. A final step is needed to unscramble the result into (L/2) - 1complex pairs, one O-frequency value, and one Nyquist-rate value. Alternatively, putting complex data such as the quantities P(s)E(-s) in this form, the inverse of the above process can be formed. Software to perform the calculations was written for this project by G. A. Morris.
The central 4096 data points of the integrated pulse profiles were selected for the templates since the correlation process involves using the FFT algorithm for 2^{12} data points. The correlation is done twice to cover all 5000 data points. The details are presented in Section V. The value of τ corresponding to the maximum value of $c(\tau)$ denotes the position of the pulse in the data array.

C. SUBTRACTION OF THE BACKGROUND

Once an estimate of the pulse phase has been made, the pulsed rower must be separated from the steady component contributed by the system. Let the signal be represented by a pulse template $U(t - \tau)$ of unity maximum height. The pulse is superimposed on a constant term N_s due to system noise. Associated with N_s are small noise variations n(t) caused by a non-zero post-detection bandwidth. Hence, the total signal S(t) becomes

$$S(t) = gU(t - \tau) + n(t) + N_{g}$$
 (8)

where g scales the template to the data. The expected value of n, denoted by $E\{n\}$, is zero. The objective is to minimize the quantity

$$I_{1} = \int \left[S(t) - gU(t - \tau) - N_{g} \right]^{2} dt$$
(9)

All terms in I₁ are constant except the term

$$2g \int S(t) U(t - \tau) dt$$

which must be maximized to minimize I1. But this is merely the cross correlation performed above.

Estimates of g and N are needed to separate the pulse from receiver noise. The estimates should minimize I_1 :

$$\frac{\partial I_1}{\partial g} = 0 = -\int S(t) U(t - \hat{\tau}) dt + g \int U^2(t - \hat{\tau}) dt + N_s \int U(t - \hat{\tau}) dt$$

$$\frac{\partial I_1}{\partial N_s} = 0 = -\int S(t) dt + g \int U(t - \hat{\tau}) dt + N_s \int dt$$

where $\hat{\tau}$ is the estimate of τ from the correlation procedures. Solving for the estimates \hat{g} and \hat{N}_s of g and N_s , respectively.

$$\hat{\mathbf{g}} = \left[\int dt \int \mathbf{S}(t) \ \mathbf{U}(t - \hat{\tau}) \ dt - \int \mathbf{S}(t) \ dt \int \mathbf{U}(t - \hat{\tau}) \ dt \right] / \mathbf{D}$$

$$\hat{\mathbf{N}}_{\mathbf{g}} = \left[\int \mathbf{S}(t) \ dt \ \int \mathbf{U}^{2}(t - \hat{\tau}) \ dt - \int \mathbf{S}(t) \ \mathbf{U}(T - \hat{\tau}) \ dt \int \mathbf{U}(t - \hat{\tau}) \ dt \right] / \mathbf{D}$$
(10)

where

$$D = \int dt \int U^{2}(t - \hat{\tau}) dt - \left[\int U(t - \hat{\tau}) dt \right]^{2}$$

Note that only the quantities involving S(t) need to be calculated for each data record. All other terms are calculated once for each template.

The discrete version of Eq. (10) is used in the data reduction. The value \hat{N}_{s} is subtracted from each data record before adding the record to the evolving template.

D. THE TEMPLATES

Several hundred thousand pulses were superimposed as described above to form the final estimates of the average pulse shapes of 24 pulsars. The results appear in Figures 6 through 29. Part (a) of each figure represents the template used in the data reduction. The abscissa presents time in terms of the pulse period P, while the ordinate is in arbitrary units of power. The vertical marks above and below the main pulse define the time origin. Some templates were smoothed beyond the

amount inherent in the receiver. The smoothing was obtained by convolving a sinc(t) function of the appropriate width with the original data. The amount of smoothing noted in the figure caption represents the interval from the peak to the first zero of sinc(t). A quantitative discussion of the choice of smoothing is presented in the next section.

Part (b) of each figure presents the details of the pulse on an expanded time scale. If the data has been smoothed more than in part (a) of the figure, the amount of smoothing is noted. The position of the pulse relative to the vertical marks above and below the main pulse defines, for that pulsar, the "zero phase" condition relative to a given epoch; i.e., to obtain a zero time-of-arrival residual relative to a given epoch, the vertical marks must be coincident with that epoch.

E. TEMPLATE SMOOTHING

The accuracy with which one can estimate the arrival time of a pulse increases as the rendom fluctuations in the signal decrease and as the sharpness of the pulse increases. However, attempts to smooth the signal to lower the effects of receiver noise also tend to destroy the sharpness of the pulse. A compromise must then be reached. This is done by estimating the uncertainty in the arrival time measurement when the signal-tonoise ratio is large.

Imagine that having performed the correlation process to measure the arrival time of the pulse, an estimate $\hat{\tau}$ has been obtained. The scaling of the template, \hat{g} , and the constant component, \hat{N}_{g} , have also been estimated. Allowing for small errors in the estimates, the model of Eq. (8) can be rewritten as a Taylor expansion:

$$S(\tau) = \hat{g}U(\Delta \tau) + \Delta gU(\Delta \tau) - \hat{g}\Delta \tau U'(\Delta \tau) + \hat{N}_{s} + \Delta N_{s} + n(\Delta \tau)$$
(11)









32

- -----

State of a state of



Ξ.

Figure 8. The template of PSR 0355+54: (a) P = 0.156 sec. The pulse has been smoothed by a window 160 µsec wide. (b) Expanded time scale

and the set of a set of the set o

· • · • ·

т. Ж

ವರಿಸ್ತಿನ ಎಂದು ಸಾಹಿತಿಯ ಕಾರ್ಯಕ್ರಿಯನ್ನು ಎಂದಿಗಳು ಎಂದಿಗಳು ಹಿಂದಿಕೆ 🎾 ಪ್ರತಿಕರ್ಷಕರಿಗೆ ಪ್ರಕರ್ಷಿಗಳು ಕಾರ್ಕಿಕ್ಸ್ ಹಿಗೆ

1.4.2.4

ź



Figure 9. The template of PSR 0525+21. (a) P = 3.745 sec. The pulse has been smoothed by a window 2.2 ms wide. (b) Expanded time scale



ł

Figure 10. The template of PSR 0628-28: (a) P = 1.244 sec; (b) Expanded time scale



Figure 11. The template of PSR 0736-40: (a) P = 0.374 sec; (b) Expanded time scale









ŝ

-175











Figure 16. The template of PSR 1237+25: (a) P = 1.382 sec; (b) Expanded time scale

÷

י ל

.7 |







à

ł

,

Star Allena







38







s;



ţ

Figure 22. The template of PSR 1911-04: (a) P = 0.826 sec; (b) Expanded time scale



÷....

Figure 23. The template of PSR 1929+10: (a) P = 0.227 sec; (b) Expanded time scale

40

5 5 18 city 1



14 P.

.

815 YEAN 1287

تە 14.

Ì

Figure 24. The template of PSR 1933+16: (a) P = 0.359 sec; (b) Expanded time scale



Figure 25. The template of PSR 2016+28: (a) P = 0.559 sec; (b) Expanded time scale



. * *

.....

ż

ŝ

 i_{ℓ}

. . .





2

.



42

N N'S A KAR F ...

ŧ



á

l

1

Figure 29. The template of PSR 2217+47: (a) P = 0.538 sec; (b) Expanded time scale, with the data smoothed for display by a window 1 ms wide

where the prime denotes a derivative, and

. . . .

$$\Delta g = g - \hat{g}, \ \Delta \hat{N}_{s} = N_{s} - \hat{N}_{s} \text{ and } \Delta \tau = \tau - \hat{\tau}$$

Substituting Eq. (11) into Eq. (9) yields the mean square error in using estimates \hat{g} , \hat{N}_{s} and $\hat{\tau}$ instead of the mean values g, N_{s} and τ . Minimize the mean square error with respect to $\Delta \tau$ to obtain

$$\frac{\partial I_1}{\partial^{\Lambda_{\tau}}} = 2 \int_0^P \left[\Delta g U(\Delta \tau) - \hat{g} \Delta \tau U^*(\Delta \tau) + \Delta N_g + n(\Delta \tau) \right]$$
$$\chi \left[\Delta g U^*(\Delta \tau) - \hat{g} U^*(\Delta \tau) - \hat{g} \Delta \tau U^{**}(\Delta \tau) + n^*(\Delta \tau) \right] d\tau = 0$$

where

U' = dU/dt, $U'' = d^2 U/dt^2$, etc. The integration is over one full pulse period. Collecting terms, we make use of the following facts:

(1)
$$U(0) = U(P) = 0$$

(2) $U'(0) = U'(P) = 0$
(3) $\int n' d\tau = n(0) - n(P) \approx 0$

and

(4)
$$\int n n' d\tau \approx 0$$

Then

$$\Delta \tau \approx \frac{\int U'n \, d\tau}{\hat{g} \int (U')^2 \, d\tau}$$
(12)

للمعيدية والارتجاب والمعيدية

where terms of the order $\Delta g/g$ have been omitted.

Since

$$E\{\Delta\tau\} = \int U' E\{n\}d\tau/\hat{g} \int (U')^2 d\tau = 0,$$

then

$$\sigma_{\tau}^{2} = E\{\Delta\tau^{2}\} = \frac{E\left\{\int U'(s) n(s) ds \int U'(q) n(q) dq\right\}}{\widehat{g}^{2} \left[\int (U')^{2} d\tau\right]^{2}}$$

or

$$\sigma_{\tau}^{2} = \frac{\iint U'(s) \ U'(q) \ R_{n}(s-q) \ ds \ dq}{\widehat{g}^{2} \left[\int (U')^{2} \ d\tau \right]^{2}}$$

where $R_n(\Delta \tau)$ is the auto-correlation function of the random fluctuation n(t). $R_n(0) = \sigma_n^2$. If R_n always decreases significantly before U' can change significantly, then this result remains independent of the post-detection bandwidth. Measuring the variance of the fluctuations in a region not containing the pulse, we can write

$$\sigma_{\tau} \approx \frac{\sigma_{n}/\hat{g}}{\left[\int_{0}^{P} (U')^{2} d\tau\right]^{1/2}}$$
(13)

The denominator is corrected for contributions to U' from random fluctuations. The expected standard deviation as ociated with $\hat{\tau}$ is then σ_{τ} .

The expression for σ_{τ} satisfies our intuition with respect to how well τ can be measured for a given signal-to-noise ratio (\hat{g}/σ_n) and a given pulse sharpness (measured by $\int (U^{\dagger})^2 d\tau$). A test of Eq. (13) was performed which dia not rely on intuition. The extremely noise-free template for PSR 0833-45 was used as the basis for producing 100 records of simulated data. A random number generator was used to superimpose uncorrelated fluctuations onto the original template. Equation (13) was

used to choose the variance σ_n^2 given that $\sigma_{\tau} = 5 \ \mu sec$. The sharpness $\int (U')^2 d\tau$ is known from the template, and g = 1 in this case.

Would the correlation method define a random process $\hat{\tau}$ from the 100 simulated data records wich the correct standard deviation? It did. The true value of τ was known in this test, and the standard deviation of the estimates $\hat{\tau}$ about the expected value τ was 5.2 µsec.

The templates were smoothed to the resolution where σ_{τ} stopped decreasing due to a lower σ_{n} and began increasing due to loss in sharpness of the pulse.

F. CENTRAL SPIKES

Occasionally, the pulse strength in some of the weaker pulsars ebbs to the point of invisibility (a S/N ratio of 1 or less), even after an hour of integration. In these data records the template seeks out the stronger, sharper random fluctuation. The resulting estimate of τ represents the position in the data array of one of the larger noise spikes. In superimposing data records to produce a template, a narrow central spike was formed.

The central spike phenomenon was observed in the templates for PSR 1818-04 and PSR 2111+46. The spike was removed by placing in the central nine array locations a random process with a mean computed from the eight neighboring values. The variance was set equal to that of the pulse-free portion of the array.

G. DEFINING THE ZERO-LEVEL

Defining the zero-level of the pulsed radiation is straightforward in the cases where the pulses are narrow and confined to one window. This is not to say that the radiation between the pulses is not in part pulsed and associated with the pulsar. In most cases, however, the

46

emission is featureless. If a portion of the flat interpulse region is also pulsed, measurements of a different nature are required to detect this possible situation. In general, the mean value of the interpulse emission is used to define the zero-level.

In several cases, such as PSR 0950+08, the presence of sharp or broad features requires a careful definition of the interpulse region. In these carefully defined regions the mean value is then taken to be the zero-level of pulsed emission.

47

۶,

1.94

SECTION V

ESTIMATING THE PULSE ARRIVAL TIME

Each data record on magnetic tape presents an opportunity to measure the arrival time of a received pulse relative to a given epoch. The measurement is completed by correlating the appropriate template (Figures 6 through 29) with data records represented by Figure 1. The correlation process leads to an estimate of the maximum cross-correlation coefficient. However, the coefficients occur only every $P_a/5000$ seconds of delay, so interpolation in the neighborhood of the largest coefficient is required to obtain the best estimate of the delay corresponding to the true maximum of the coefficients. The value of the largest correlation coefficient and an estimate of the random fluctuations due to receiver noise are used to estimate the uncertainty in the time delay.

A. THE CORRELATION PROCESS

The data records are 5000 words long. Two arrays of 4096 words each are constructed to allow easy use of the FFT algorithm. The two separate results, each 4096 words, are combined to yield one 5000-word array of cross-correlation coefficients.

The template is shifted in position before transformation, as shown in Figure 30. The time within each pulse corresponding to <u>the</u> <u>arrival time of the pulse</u> is, in practice, the time marked "O" in Figures 6 through 29. Each template is shifted circularly until this time occurs in the first word of each array. Hence, the position of the largest correlation coefficient will be the estimate of the arrival time of that particular pulse. In addition, this shift keeps the size of the imaginary component of the transformed template to a minimum.



Figure 30. Shifting and Transforming the Template

ęį.

ł

obtained when no smearing is present represents a bias that can be estimated and removed. The bias is estimated by first calculating the expected arrival time based on the model of the pulsar, and then comparing that with the arrival time expected for a pulsar with a constant period and constant Doppler shift. The mean of the differences is the estimate of the bias.

The expected arrival time t_k of the $k^{\mbox{th}}$ pulse arriving after the starting epoch t_0 is given by

$$t_{k} = t_{bk} + \frac{1}{c} \left[d(t_{k}) - d(t_{0}) \right]$$
(15)

where t_{bk} is the elapsed time between k pulses at the barycenter of the solar system, d(t) is the distance of the received wavefront from the barycenter, and c is the velocity of light. Expanding the barycentric pulse period P(t) in a Taylor series about t_0 ,

$$P = P_0 + \dot{P}_0(t - t_0) + \ddot{P}_0(t - t_0)^2/2 + 0(\ddot{P} t^3)$$

It can then be shown that

$$t_{bk} \approx k P_0 + k^2 P_0 \dot{P}_0 / 2 + 0 (k^3 P)$$
 (16)

The quantity d(t) can be replaced with the scalar product $-\overline{r}_{a}(t)$ $\cdot \overline{U}_{p}(t)$, where $\overline{r}_{a}(t)$ is the position vector of the antenna relative to the barycenter at time t. The unit vector \overline{U}_{p} is the direction vector of the pulsar as seen from the barycenter. Referring to Figure 34, note that $|\overline{r}_{a}| = a \sim 1$ AU, and R/a >> 1. Using the facts that

 $2(R-d)Y = x^2 + y^2$

and

 $x^{2} + y^{2} = z^{2} = d^{2} + a^{2} + 2ad \cos \theta$



Figure 31. Computing the Cross Correlation Coefficients



Figure 32. An Example of the Cross Correlation Process; PSR 1818-04 on 27 Jan. 1970 at 16^h 58^m 00^s UT

B. ESTIMATING THE DELAY

The correlation coefficients are examined to determine the position in the array of the largest coefficient. In the cases of PSR 0355+54 and PSR 0833-45, two and three pulses, respectively, occur per data record. In these cases two or three positions are determined, each separated by a pulse period, and averaged together at the end of the estimation process.

Subsequent refinement of the estimate of the pulse position is obtained through interpolation. Interpolation is obtained by fitting, in a least-square sense, a fourth-order polynomial to M coefficients either side of the largest coefficient. At position x_1 (measured relative to the position of the largest coefficient) in the array of $\overline{}$

correlation coefficients, a predicted value of $C_i = \sum_{\ell=0}^{4} a_{\ell} x_i^{\ell}$ is obtained. Note that $x_{i+1} - x_i = 1$. We have for the 2M + 1 coefficients involved,

$$C_{-M} = a_{0} + a_{1}x_{-M} + \dots + a_{4}x_{-M}^{4}$$

$$\vdots$$

$$C_{0} = a_{0}$$

$$C_{1} = a_{0} + a_{1}x_{1} + \dots + a_{4}x_{1}^{4}$$

$$\vdots$$

$$C_{M} = a_{0} + a_{1}x_{M} + \dots + a_{4}x_{M}^{4}$$

The polynomial coefficients a_j need to be determined. Denoting the measured coefficient by \hat{C}_i , we want to minimize the quantity

$$s = \sum_{i=-M}^{M} (\hat{c}_{i} - c_{i})^{2} = \sum_{i} \left[\hat{c}_{i} - \sum_{\ell=0}^{4} a_{\ell} x_{i}^{\ell} \right]^{2}$$

Requiring $\frac{\partial S}{\partial a_{l}} = 0$ for l = 0 to 4, we find the set of equations

$$\begin{aligned} & \mathfrak{l} = 0; \ \sum_{i=-M}^{M} \hat{c}_{i} = a_{0} + a_{1} \sum_{i} x_{i} + a_{2} \sum_{i} x_{i}^{2} + a_{3} \sum_{i} x_{i}^{3} + a_{4} \sum_{i} x_{i}^{4} \\ & \mathfrak{l} = 1; \ \sum_{l=-M}^{M} \hat{c}_{i} x_{i} = a_{0} \sum_{i} x_{i} + a_{1} \sum_{i} x_{1}^{2} + \dots + a_{4} \sum_{l} x_{i}^{5} \end{aligned}$$

$$(14)$$

$$\ell = 4; \sum_{i=-M}^{M} \hat{c}_{i} x_{i}^{4} = a_{0} \sum_{i} x_{i}^{4} + a_{1} \sum_{i} x_{i}^{5} + \dots + a_{4} \sum_{i} x_{i}^{8}$$

Since the x_i occur at uniform intervale, the sums of odd powers of x_i are zero, and the sums of even powers of x_i are precomputed for each pulsar. The size of M is chosen separately for each pulsar. Within the range $x_{-M} < x_i < x_M$, the fourth-order polynomial provides a good fit to the correlation coefficients obtained when the template is correlated with itself.

The system of equations in Eq. (14) is solved for the polynomial coefficients a_{l} . The curvature of the resulting polynomial is computed. A positive second derivative indicates a distortion due to a low signal-to-noise ratio, and the interpolation is not attempted. Otherwise, the position x_{p} at which the polynomial peaked is determined. Then, stepping in intervals of 0.01 between x_{p} -1 and x_{p} +1, the refined estimate x_{d} of where the polynomial reaches its peak is determined.

The arrival time t_a of the first pulse occuring after the start time t_s is readily calculated. The separation between data array elements is $P_a/5000$ seconds. Hence, $t_a = t_s + x_d P_a/5000$.

The interpolation performed near the peak coefficient in Figure 32 is presented in Figure 33. The measured values of the cross-correlation coefficients (solid circles) are displayed versus the corresponding array location, which is proportional to time delay. The solid curve is the polynomial resulting from the least-squares fit. The pca. of the polynomial is near x = 589.0. The region near this value of x is enclosed by the broken box, and reproduced on a finer scale in the inset. Progressing from x = 589 to x = 590 in steps of 0.01, the peak, denoted by the vertical arrow, is found at $x_d = 589.79$.



Figure 33. Correlation Coefficients Versus Time Delay Near the Peak Shown in Figure 32

C. ADJUSTMENTS

If the template has been shifted due to a large asymetry about the zero-phase position, the effect of that shift is removed at this point in the data reduction. If two or three estimates of x_d were made, they are averaged together at this point.

D. UNCERTAINTY IN THE ARRIVAL TIME

Each estimate of the arrival time $\hat{\tau}$ is accompanied by an estimate of the uncertainty σ_{τ} . A modification of Eq. (13) is used. The scale factor \hat{g} and the baseline level \hat{N}_{g} are first estimated using Eq. (10). Quantities dependent on the template U(τ) only, having been computed earlier in the template generating process, are on magnetic tape with

the template itself. Note that the quantity $\int S(t)U(t - \hat{\tau}) dt$ is merely the maximum correlation coefficient. The root-mean-square (rms) noise level σ_n includes a contribution by the template. The denominator of Eq. (13) is another quantity computed earlier.

1

 \mathcal{T}_{i}

· · · · · ·

SECTION VI

GEOCENTRIC ARRIVAL TIMES

Now that the arrival time of the pulse has been measured relative to an epoch defined by a particular clock, it is time to consider referring these station arrival times to the geocenter. There are several corrections which must be applied to the measured arrival time before useful geocentric results are obtained: (a) A bias in the arrival time, caused by the smearing of the pulse during superposition, must be estimated and removed; (b) the geometric correction, changing the spatial reference from the antenna to the geocenter, must be applied; (c) an extra delay must be removed which is caused by using frequencies lower than 2388 MHz; (d) a small annual effect caused by the interaction of the Doppler and dispersion phenomena must be removed; (e) the clock epochs used in the measurements must be expressed as Coordinated Universal Time (UIC) epochs; (f) UTC epochs must then be expressed as ephameris time (ET) epochs. The result is then the arrival time that would have been obtained had the measurement been performed at the geocenter using a clock running on ephemeris time.

A. CORRECTION FOR PULSE SMEARING

A small amount of smearing is inevitable as the received pulses are superimposed. The Doppler shift caused by Earth's rotation and orbital motion changes over the interval of integration. Also, differences can exist between the actual pulse period and the assumed period. The result is a slow drift in the pulse phase relative to the starting epoch, causing the centroid of the integrated signal to drift forward or backward in time as pulses are superimposed. The difference between the phase of the centroid of the superimposed pulses and the phase

57

「そび治死」み、

obtained when no smearing is present represents a bias that can be estimated and removed. The bias is estimated by first calculating the expected arrival time based on the model of the pulsar, and then comparing that with the arrival time expected for a pulsar with a constant period and constant Doppler shift. The mean of the differences is the estimate of the bias.

The expected arrival time t_k of the $k \stackrel{th}{\cdot}$ pulse arriving after the starting epoch t_0 is given by

$$t_{k} = t_{bk} + \frac{1}{c} \left[d(t_{k}) - d(t_{0}) \right]$$
(15)

where t_{bk} is the elapsed time between k pulses at the barycenter of the solar system, d(t) is the distance of the received wavefront from the barycenter, and c is the velocity of light. Expanding the barycentric pulse period P(t) in a Taylor series about t_0 ,

$$P = P_0 + \dot{P}_0(t - t_0) + \ddot{P}_0(t - t_0)^2/2 + 0(\ddot{P} t^3)$$

It can then be shown that

$$t_{bk} \approx k P_0 + k^2 P_0 \dot{P}_0 / 2 + 0 (k^3 P)$$
 (16)

The quantity d(t) can be replaced with the scalar product $-\overline{r}_{a}(t)$ $\cdot \overline{U}_{p}(t)$, where $\overline{r}_{a}(t)$ is the position vector of the antenna relative to the barycenter at time t. The unit vector \overline{U}_{p} is the direction vector of the pulsar as seen from the barycenter. Referring to Figure 34, note that $|\overline{r}_{a}| = a \sim 1$ AU, and R/a >> 1. Using the facts that

 $Y = -a \cos \theta - d$

 $2(R-d)Y = X^2 + Y^2$

and

$$X^{2} + Y^{2} = Z^{2} = d^{2} + a^{2} + 2ad \cos \theta$$





the distance d becomes

ť,

 $d \approx -a \cos \theta = -\overline{r}_a \cdot \overline{U}_p$

where R/a >> 1. (The smallest value of this ratio is about 10^7 .)

Consider the arrival time obtained if $P(t) = P_a$ over the entire interval of integration. Suppose there is a delay of ℓ pulses from the time the timing pulses are started to the time when the data collection is started (usually a few seconds at most). Then M groups of N pulses (usually 500) are superimposed, with n pulses elapsing between groups to allow for a visual display of the progress of the observation. Figure 35 displays the numerology used in counting these pulses. Under the

59

States and a



Figure 35. Nomenclature for Enumerating Pulses

assumption of a constant period, the arrival time $t_{ak}^{}$ of the k^{th} pulse after the start epoch $t_0^{}$ is

$$t_{ak} = [l + j + (N + n)(m - 1)] P_a = kP_a$$

where j is the pulse in question in the mth block of N pulses, and P_a is the assumed constant period. (The synthesizer frequency f_s used in the observation, as per Figure 2, is $10^6/P_a$.)

The mean of the difference $t_k - t_{ak}$ is given by

$$E\{t_{k} - t_{ak}\} = \overline{t_{k} - t_{ak}} = \frac{1}{NM} \sum_{m=1}^{M} \sum_{j=1}^{N} (t_{k} - kP_{a})$$

Let $D = \frac{t_k - t_{ak}}{ak}$, representing the bias caused by the smearing. Substituting Eqs. (15) and (16) into the expression for D yields

$$D = \frac{(P_0 - P_a)}{NM} \sum_{m=1}^{M} \sum_{j=1}^{N} \{j + \ell + (N + n)(m - 1)\} + \frac{P_0 P_0}{2NM} \sum_{m=1}^{M} \sum_{j=1}^{N}$$
(17)

5

$$\{j + l + (N + n)(M - 1)\}^{2} + \frac{1}{cNM} \sum_{m=1}^{M} \sum_{j=1}^{N} \{d(t_{k}) - d(t_{0})\}$$

The mean of the distances $d(t_k)$ can be replaced by the distance near the midpoint of the range of j, yielding for the third term of Eq. (17),

$$-\frac{1}{cM}\sum_{m=1}^{M}\left\{\overline{r}_{a}\left[\frac{t_{N+1}}{2}+\ell+(N+n)(m-1)\right]-\overline{r}_{a}(t_{0})\right\}\cdot\overline{v}_{p}(t_{0})$$

Performing the indicated summations, Eq. (17) becomes

$$D = \left[\frac{N+1}{2} + i + \frac{(N+n)(M-1)}{2}\right] \quad (P_0 - P_a)$$

$$+ \frac{P_0 \dot{P}_0}{2} \left\{ \left[\frac{N+1}{2} + i + \frac{(N+n)(M-1)}{2}\right]^2 + \frac{(N^2 - 1) + (N+n)^2 (M^2 - 1)}{12} \right\}$$

$$- \frac{1}{cM} \sum_{m=1}^{M} \left\{ r_a \left[\frac{t_{N+1}}{2} + i + (N+n)(m-1)\right] - \overline{r}_a(t_0) \right\} \cdot \overline{U}_p(t_0)$$
(18)

It has been assumed that the signal strength is constant over the range of k. Large fluctuations cause Eq. (18) to be in error, and if the drift is large, the results may be rejected.

The expression derived above for the drift is applied to all the data. In all cases, N = 500. The delay l is usually 0. The plot time n is equivalent to 30 seconds before Julian date 2440195.5, and 17 seconds thereafter. The number M of 500-pulse integrations varies depending on pulse strength, ranging from the equivalent of a few minutes to over two hours.

The Jet Propulsion Laboratory ephemeric DE 96 is used to compute bar; entric positions \overline{r}_a . One major export from this observing program is ephemeric independent, geocentric arrival times. However, the drift correction is needed, and is best done by the observer early in the reduction process. The use of a different ephemeric of similar quality

will not change the corrections of Eq. (18) significantly. A large drift correction is on the order of a few milliseconds, and a velocity difference between two ephemerides of greater than 0.1 percent would be required to produce a 1 µsec change in the drift estimate. This is usually less than the error estimates encountered in these measurements. Note also that this correction is dependent on a good knowledge of the pulse period and its derivatives. These have been well-determined in a separate data analysis effort.

The sum of the barycentric position vector of the geocenter and the geocentric position of the antenna, $\overline{r_g}$, yields $\overline{r_a}$. The computation of $\overline{r_g}$ is discussed in the next section. The direction vector $\overline{U_p}$ is derived from a model-fitting procedure, which is beyond the scope of this report. All the pulsar positions used here are known to at least 1 arc-sec, an error that will produce a geocentric arrival time error of 0.1 µsec or less.

An example of phase drift and the subsequent correction for the effect is shown in Figure 36. Sixteen groups of 500 pulses of PSR 0833-45 were examined separately to estimate τ . The solid circles represent these estimates relative to the estimate for m = 1. The line through the solid circles represents a fit-by-eye to the linear drift. The triangle is the result of examining the accumulation of the 16 blocks of data. As one would expect in a well-behaved situation, this estimate of τ has drifted about one-half the distance from "0" as the estimate for the m = 16 case alone. This line through the triangle and the zero for this example (0 at m = 1) delineates the hypothetical drift one would find in examining ar immulation of the preceding blocks of data. The quantity D, derived from Eq. (18), is the amount the result represented



Figure 36. An Example of Drift in the Phase of PSR 0833-45 on 1 Dec. 1968 at 11h 00m 00s (UT)

by the triangle is shifted to correct for drift. Note that by shifting the datum down by D ms, it is nearly in line with the intersection of the dashed line and the m = O axis. In a later analysis the resulting $\hat{\tau}$ was found be only 15 used away from the arrival time predicted by a smooth model, which was fitted to several months of data.

B. ANTENNA POSITION

The referral of the measured arrival time to the geocenter requires the addition of the time delay $\overline{r}_g \cdot \overline{U}_{pg}/c$, where \overline{r}_g is the geocentric position of the antenna and \overline{U}_{pg} is the geocentric direction vector of the pulsar. In practice $\overline{U}_{pg} \approx \overline{U}_p$. The coordinate system used here is the right-handed rectangular system (X, V, Z). The coordinate X is in the direction of 0^h Right Ascension (FA) at 1950.0. The station position at a particular UTC cpech is calculated from the ante-na coordinates in

Table 5 and the knowledge of the Greenwich Hour Angle (GHA). The station coordinates are a version of Location Set (LS) 44, which are based on the ephemeris DE 96 (Ref. 19). The GHA is computed as outlined in Table 6 (see Ref. 20).

Antenna	West Longitudeβ (deg)	Latitude λ (deg)	Radius R (km)
DSS 11 (26m)	116.849390	35,208047	6372.011
DSS 12 (34m)	116.805462	35.118665	6371.994
DSS 13 (26m)	116.794863	35.066546	6372.117
DSS 14 (64m)	116.889507	35.170847	6366.227
DSS 62 (26m)	4.367811	40.263145	6369.963
DSS 63 (64m)	4.247991	40.241318	6370.048

Table 5. Geocentric Antenna Coordinates

Table 6. Relations Used in Computing the GHA of an Observing Station

```
At the epoch d + SEC/86400,

(3HA = 100.0755426 + d(0.985647346 + 2.9015 x 10^{-13} d)

+ \dot{0} SEC + \delta\Psi cos(\epsilon) degrees

where: (1) d = Julian days past 2433282.5

(2) SEC = seconds past 0^{h} on day d

(3) \dot{0} = 4.17807417 x 10^{-3}/(1+5.21 \times 10^{-13} \text{ d}) deg/sec

(4) \delta\Psi, the nutation in longitude, is taken from

ephemeris DE 96

(5) \dot{\epsilon} = 4.09205684 x 10^{-1}

+ T[-2.28534 x 10^{-4} + T(-1.5 x 10^{-8} + 8.7 x 10^{-9})] radians

where T is in Julian centuries of 36525 days past

2433282.5.
```
The components of the geocentric antenna vector are computed in the 1950.0 coordinate system, requiring rotations for precession and nutation. Considering an antenna at longitude β and latitude λ at a given epoch,

$$\overline{\mathbf{r}}_{g} = \begin{bmatrix} \mathbf{r}_{gx} \\ \mathbf{r}_{gy} \\ \mathbf{r}_{gz} \end{bmatrix} = \begin{bmatrix} P_{11} & P_{12} & P_{13} \\ P_{21} & P_{22} & P_{23} \\ P_{31} & P_{32} & P_{33} \end{bmatrix} \chi \begin{bmatrix} 1 & -\delta\Psi\cos(\overline{\epsilon}) & -\delta\Psi\sin(\overline{\epsilon}) \\ \Delta\Psi\cos(\overline{\epsilon}) & 1 & -\delta\epsilon \\ \Delta\Psi\sin(\overline{\epsilon}) & \delta\epsilon & 1 \end{bmatrix}$$

$$\chi \begin{bmatrix} \mathbf{r}'_{gx} \\ \mathbf{r}'_{gz} \\ \mathbf{r}'_{gz} \end{bmatrix}$$

where

 $\begin{bmatrix} \mathbf{r'}_{gx} \\ \mathbf{r'}_{gy} \\ \mathbf{r'}_{gz} \end{bmatrix} = \begin{bmatrix} \operatorname{Rcos}(\lambda)\operatorname{cos}(\operatorname{GHA} - \beta) \\ \operatorname{Rcos}(\lambda)\operatorname{sin}(\operatorname{GHA} - \beta) \\ \operatorname{Rsin}(\lambda) \end{bmatrix}$

The elements P_{ij} of the precession matrix, given in Table 7, are f Ref. 21. An approximation to the nutation matrix has been used, which $\delta \Psi$ and $\delta \varepsilon$ are taken from the ephemeris DE 96, while $\overline{\varepsilon}$ is computed as in Table 6.

The approximations to the precession and nutation matrices are accurate to a few parts in 10^9 in both angular position and radial magnitude. Since $|\overline{r}_g| \sim 6 \times 10^3$ km, the corresponding error in arrival time is on the order of picoseconds.

0.5

1. 14+4

Table 7. Elements of the Precession Matrix

$$P_{11} = 1.0 - T^{2}(2.9697 \times 10^{-4} + 1.3 \times 10^{-7}T)$$

$$P_{12} = T \left[2.234988 \times 10^{-2} + T(6.76 \times 10^{-6} - 2.21 \times 10^{-6} T) \right]$$

$$P_{13} = T \left[9.71711 \times 10^{-3} - T(2.07 \times 10^{-6} + 9.6 \times 10^{-7}T) \right]$$

$$P_{21} = -P_{12}$$

$$P_{22} = 1 - T^{2}(2.4976 \times 10^{-4} + 1.5 \times 10^{-7} T)$$

$$P_{23} = -T^{2} (1.085 \times 10^{-4} + 3 \times 10^{-8}T)$$

$$P_{31} = -P_{13}$$

$$P_{32} = P_{23}$$

$$P_{33} = 1 - T^{2} (4.721 \times 10^{-5} - 2 \times 10^{-8}T)$$

Note: T is in Julian centuries of 36525 days past 2433282.423.

Larger errors appear when referring the arrival times to the solar system barycenter. Since one normally works in 1950.0 coordinates and solves for pulsar positions and proper motions in these coordinates, barycentric arrival times are not highly sensitive to the recent increase in the accepted value of the general precession in longitude by 1.1 arc-sec/ century. However, changes in the Earth's barycentric position can vary significantly as ephemerides are improved. For example, the difference in the angular position of Earth in DE 108 and that in DE 96 adds up to

66

1.14.1

about 0.5 arc-sec/century.* In a ten-year observing program, the difference is then 0.05 arc-sec, similar to the errors in pulsar positions derived from the arrival time data.

C. DISPERSION

Considerable effort was expended in the early observations to obtain one frequency, 2388 MHz. This was not always possible, for several observations occurred at 2295 and 2328 MHz. All observations are now at 2295 MHz. Integration of Eq. (1) provides the corrections in time to be added to an arrival time measured at F MHz to refer it back to 2388 MHz. Performing the integration, the correction ΔT_D becomes

$$\Delta T_{\rm D} = \frac{\rm DM}{2.410086 \times 10^{-4}} \left[\frac{1}{(2388)^2} - \frac{1}{\rm F^2} \right] \ \rm sec$$
 (19)

where the values of DM are from Table 4.

In the application of ΔT_D to PSR 0833-45 it was found that an add.:ional 250 µsec had to be subtracted from the measurement when F = 2295 MHz. No significant addendum was needed when F = 2328 MHz. The average pulse shape must be independent of observing frequency if the correction ΔT_D is to be perfectly adequate. Most shapes do change significantly with frequency, including that of PSR 0833-45, thus creating the need to measure an addendum to ΔT_D .

D. DOPPLER DISPERSION

The actual frequency of observation varies with the component of antenna velocity laying along the direction of the pulsar. The measured arrival time then has a varying component since the interstellar medium

^{*}E.M. Standish, private communication.

is dispersive. The Doppler-shifted observing frequency F_0^{\dagger} is related to the mean observing frequency F_0^{\dagger} by

$$\mathbf{F}_{0}' = \mathbf{F}_{0} \quad \left(1 + \frac{\overline{\mathbf{V}}_{a} \cdot \overline{\mathbf{U}}_{p}}{c}\right)$$

where \overline{V}_a is the barycentric velocity of the antenna. In this situation, integration of Eq. 1 leads to a form similar to Eq. (19). Performing that operation,

$$\Delta T_{DD} = \frac{DM}{1.205043 \times 10^{-4} F_0^2} \begin{bmatrix} \frac{1}{\overline{v} \cdot \overline{v}} & -1 \\ \frac{1}{\overline{v} \cdot \overline{v}} & -1 \\ 1 + \frac{a}{c} \end{bmatrix}$$

where ΔT_{DD} , the correction for Doppler dispersion is added to the measured arrival time. Using the fact that $|\overline{V}_a/c| < 10^{-4}$, and setting $F_0 = 2388$ MHz, the correction becomes

$$\Delta T_{\rm DD} \approx 1.455220 \text{ x } 10^{-3} \text{ DM} \left(\overline{V_a} \cdot \overline{U}_p\right) /c$$
(20)

The magnitude of this yearly oscillation is about 6 μ sec for PSR 0833-45 and about 18 μ sec for PSR 1933+16. This correction is not small in all cases compared to the measurement accuracy, nor is it small compared to the desired uniformity of the approximation to ephemeris time.

E. ARRIVAL TIME IN UTC

.....

. 4

The epoch at which the data sampling process begins is defined by a 1-sec pulse from the station clock. A sequence of 1-second pulses (clock pulses) is sent continuously via microwave link from the Standards Laboratory at the Goldstone complex to the individual tracking stations. Allowing for the delay over the links, these pulses define a U/C which is within 5 µsec of UTC as disseminated by the National Bureau of C,

Standards (NBS). The 1-second pulses generated at DSS 14, UT(DSS 14), are kept within 5 µsec of the Standards Laboratory pulses. Therefore, at DSS 14 the 1-sec pulses are assumed to be within 10 µsec of UTC (NBS). The 1-sec pulse used at DSS 13, UT(DSS 13), is frequently synchronized with the Standards Laboratory pulses to within 1 µsec. A piece-wise linear curve approximating the deviation in this 1-sec pulse from UTC(NBS) is used to correct epochs at DSS 13 to UTC (NBS). The correction is at most 100 µsec.

The correction equal to the width of the clock pulse at DSS 14 is subtracted as part of this correction. (See the discussion of system codes 2, 3, and 5, and Fig. 3.) The small correction equal to $P_a/5000$ is then added if appropriate.

F. CONVERSION FROM COORDINATED UNIVERSAL TIME TO EPHEMERIS TIME

Coordinated Universal Time (UTC) is an approximation to a uniform time based on the rotation of Earth. Several steps are necessary to convert to the truly uniform ephemeris time (ET). Of prime importance is the correction for annual effects due to the orbital motion of the Earth-based clocks about the Sun. The correction Δ ET is given in (Ref. 16) as

 $\Delta ET = 0.001658(sin E + 0.0368) sec$ (21)

where E is the eccentric anomaly of the Earth in a heliocentric orbit. Letting M represent the mean anomaly of Earth and e the orbital eccentricity,

sin E = sin M (1 + e cos M)

Letting an ET epoch be represented by a Julian ephemeris date (JED) plus the number of seconds (SEC) past 0^{h} ET,

$$M = 358.000682 + 0.9856002628d - T^{2}(1.55 \times 10^{-4} + 3.33 \times 10^{-6}T) \text{ degrees}$$
$$e = 0.0167301085 - T(4.1926 \times 10^{-5} + 1.26 \times 10^{-7}T)$$

where

-* _{s.t}

d = (JED - 2433282.5 + SEC/86400) days

T = d/36525 centuries

(see Ref. 17)

Eq. (21) omits a monthly term of 1.5 μ sec, a 12-year term of 21 μ sec, and a 29-year term of 5 μ sec (Ref. 10).

The ET epoch corresponding to each UTC arrival time, represented by a Julian date JD plus the number of seconds SEC past 0^h , is computed from

$$ET - UTC(USNO) = \begin{cases} \Delta ET + 38.662738 + 0.002592d \text{ sec}; JD < 2441317.5 \\ (22) \\ \Delta ET + 32.184 + n \text{ sec}; JD > 2441317.5 \end{cases}$$

where ΔET is computed using Eq. (21), d = JD - 2440000.5 + SEC/86400, n is an integral number of seconds, increasing once or twice a year by 1 second (Ref. 15), and UTC(USNO) is the UTC kept by the U.S. Naval Observatory. The following assumptions and relations were used in computing Eq. (22) before 1 Jan. 1972:

(1) $|UTC(NBS) - UTC(USNO)| < 10 \ \mu sec (Refs. 12, 13)$

(2) A.1 - UTC(USNO) =
$$6.5131177 + 0.002592d$$
 sec (Ref. 14)

(3) IAT - A.1 = -0.03439 sec (Ref. 15)

(4) ET - IAT = 32.184 sec (Ref. 15)

70

1.

After Jan. 1972, the appropriate number of integral seconds is added to the UTC epoch to obtain IAT (Ref. 15) under the assumption that |UTC(NBS)- UTC (Bureau de l'Heure) < 10 µsec (Ref. 12). Then ET is obtained from ET - IAT = 32.184 sec.

G. AN EQUIPMENT IDIOSYNCRASY

The equipment problem of system code 3 requiring a correction of -0.996550 seconds is taken into account along with the corrections discussed above.

H. A TEST OF THE MEASUREMENT CONSISTENCY

The consistency of the measurements is determined by comparing the results using the procedures described earlier in the editing process (Sec. III). This comparison led to the correction of 0.996550 sec for the equipment failure, and the additional 250 µsec correction at 2295 MHz for PSR 0833-45. The question finally arises concerning the consistency of arrival times measured at different stations. Six epochs exist for PSR 0833-45 for which measurements were made at DSS 13 and DSS 14 within 1 day of each other. Results of three of these measurements showed DSS 14 results differing significantly from those at DSS 13. There were, however, serious receiver tuning problems at DSS 14 at two of the three epochs. Of the three results showing agreement, two are presented in Figure 37, where the difference (residual) between the measured and predicted arrival times is presented for each Julian date (JD) of observation (MJD = JD - 2440000.5). The agreement is remarkably good, leading to the conclusion that no systematic differences between DSS 13 and DSS 14 measurements remain in this data set.



Figure 37. Time-of-arrival Residuals of Measurements Made at DSS-13 and DSS-14 on the Modified Julian Date (MJD) of Observation for PSR 0833-45

SECTION VII

THE TABULAR RESULTS

An example of the observational results is presented in Table 8. Columns 1 and 2 list the ephemeris time of the pulse arrival at the geocenter as a Julian ephemeris date and the number of seconds past $0^{h}(\text{ET})$. (Note that no complete removal of the dispersion effect has been applied. That is, these arrival times refer to F = 2388 MHz, not F = ∞ .) The estimate of the expected standard deviation of the measurement (defining the 67 percent confidence interval if the measurement errors are Gaussian) appears in Column 3. In Column 4, all of the time conversions from UTC to ET discussed in Section VI are listed. Column 5 contains the applied drift correction, while the configuration code appears in Column 6. Barycentric arrival times are listed in Column 7 for analysts satisfied with the accuracy of DE 96.

ACKNOWLEDGEMENT

Over the course of ten years many individuals have contributed to the success of this observing program, most of them on the staff of the NASA Deep Space Network. Dedicated, excellent work was and is performed in the collection of the data by G. Birkedahl, R. Genzmer, J. W. Hudson, C. Kodak, C. Mitchell, R. McConahy, D. Rife, L. Skjerve, D. Spitzmesser, L. Wadsworth, G. Wischmeyer, and the staff of DSS 62. Much patient support has been contributed by R. M. Goldstein, R. R. Green,

E. B. Jackson and G. S. Levy.

Critical technical support was kindly provided by S. S. Brokl, C. F. Foster, R. R. Green, G. A. Morris, K. I. Moyd, A. G. Slekys, J. Urech, and M. Urech.

Table 8. PSR 0823+26 Biocentric Times-of-Arrivai

, ,

	3412 2183W3					BARYCENTRI C	-	DHEMERIS	T I ME					34440544410	
1	SEC PAST	F.		14140	2 4 60	147 145	•	SEL P		ь			540	7471774	
55	34 ET 17332.464563					71453,337786	• i		2 486.26			For the second		1175 687-945533	
			544146-94	- 1 2 8 4	20	C12499*C62**			57925		39.724421		1	6.753.956.556.6	
276	2139-922326	5	39.399323	4E4+) C	P543+224551	• •	8 58433+8	139.75		5+74070E	16-		67959+425899	
276	5439-633974	156	39-392411	SE11-	0	5945+853729		4 51243-3	56941	5	39.731195	-1	- 	62747+324929	
275 3	314683+683416	5	39+399316	646 m	0	#E2197+74C1E		1-0-1845 4	27777	11	13+75+56J	- 27	0	64347+J20024	
Etz	454277-9581	•	39+4-37353	11.	0	2214-325455	¥ •	1-04489 4	9016C	•	13.756354		5	67947-214327	
583	5433+463346	11	39+411178	26	n	4F13+432579	÷	1 51342+3)28639 	ພິ	5950CE+EE	5 1	r	53372+213517	
Egë	939+58+439	Ē	145414-66	•	C	9413+497270	- 		12250	.	536312965		0	42314•33525 ³	
5	1997/28+3681		51E/4+4E		0	446+E+E/34	•		11641/		55CG16+6E	*) I 	n		
593	5639+52353535	21	6666436E		0	5772-813610	•	1.C+96E E	98256	N 1	39+950915 52 5-55	21	0	69/1/6•1246E	
273	9239•78253N	C	33+434-12		Ô	1945CB+4266	•		111604		37+954664	2.2	0		
	744697698645	- K				1727.21212	د د ه ه					η η. Υ -	n a	6cc/cq•6546E	
297	9139-522358		10553496	, . .	00	9225+52547	in	E-C+C9E 90	91146	1	4494644	-525-	0	561014-52656	•
353 3	102623-65264	5	39.437133	i E	a	79479-212658	•	6 39643+3	37592	56	1++856+6E	9E9-	n	95556, +25556	
FECE	847568-96856	.	39.461303	27	0	81275.27544	in I	C+C+2+E E	78079		1591Cr+C+	4	n	34195 333755	
+CE	9555C6+6E5E	•	39++68532	19	0	3977-654574	۰. ۱۳	5.C+87E E	248849	-	448500+04	E:	n	37795+553333	
† CE	7533+529311	2	39+471489	2	0	7776-931364	ю. •	1.C+982 C.	1372 3 9	et)	46451C+C4	5	n	28544+534739	·
313 1	52d615+66625	27 2	29+451253	1 E	0	53224+794362		-C++2E C.	55432	22	55661C+C+	4	n	32444+553777	
116	7233+722479	5	39.492392		Ø	7423+717370		5.04885 40	223673	*	5/5/20+0+	,е С	n	28879+707049	_
324	2233•959177	195	39+517531		C	25315.465774		E+C++2E +4	1312 1212		ESETEC+C+		C	196959+81455	
332	945450-0469	5	39.545251	-158	0	6950+931192	•	* 36743+3	593296	69 9	+J+J35575	2	n	36379+759254	
332 1	10539+733435		39+5+6097	1 01-	n	10553+217559	in i •	2+C+90E 11	+CECE2	en d	116901+04		0	20179+735053	
1 ZEE	10139-693332				0 (14143-853461	*	6-C244E 19	IC/CEP	•	7564C1 %		0 0	-C+C/I+//S+F	
			3401001/40 100124100		DC	220421+CD2/		9.C+984 14		• •		1 m n n i e	n c		_
		, c	19-12-12-12 19-12-12-12) C	92792.536510	i di		940049	- C	217272715	-115) Q	32332°25°25	
345	12228-66664		39.575297	•76	n a	142604-46162	•	3 360 40 4	555029	LE LE	19:272.61	611-	0	35525.55656	
5 5+E	32939 • 585577	15	39-579184	59-	C	82733.873999		3 3964346	552398	23	40+269+83	-82	n	\$C675C+251C#	• •
94E	7239.643949	E 1	39+552161	-6-	C	7132.797299	- iv	5 - CCE61 21	46E1CS	16	\$0097E+C4	241.	n	19533+*2285	
256	75522+032979	51	39+593141	-913	0	75357+897343	•		456824	53	40-12225		റ (51657C.657ES	
C ZCE	767159+655757		39+599347		n	1546110//66/		1.00402 E		N N			n	26522 • /CEq2	
	53149.913154		39.099551	E	ററ	52977.107997 84476.929667			4EE 169	~	515644+64		0	14311.543637	
	19-4-4-4-9998		39646966	1) C	64323-543123	5 - 6 		56038				2	701672-01669	
339 7	72543+593339	•	39+409337	ÉE-	0	71923.365835		10660-7	154316	11	40.519947	-115	0	10551 - 522456	•
359 7	15339+725470	16	39+512205	EE.	C	75722+653721		8 14203-5	05820	25	+0-5160++	-97	0	14121-122434	•
359 7	19239-654476	18	39.514897	•51	0	79322+295152		2 • C++21 +	39522	*	43+523865		0	17311+294753	
365 7	F2342+23727	72	39+528315	- <u>5</u> -	C	71771.556749	к •	10 10840+3	Cesee	•	+ 3+573235	59°	n	12557+57221	
2 99E	75533+553192	191	33+641351	6 9	C)	75371+323947	*	17 8923+9	973539	5. G	+) • 5 9 1 3 7 5		n i	B515+3+3253	
385	200101.004667	м ч Ф -	346889968 345723785		00	74971.015545	•	5 1120 S	16371		# 0 # 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	52.	n (55747.465 75.35.1545	•
		5 6			 6		•	3+14867 E		n c D a			n d		
		л (т			0 C.	3/600 - 00 - 00 - 00 - 00 - 00 - 00 - 00	• •		11/10/			7 m 1 m	י ר	,	
1	********	4		1	-		•			1			כ		

ORIGINAL PAGE IS OF POOR QUALITY

•

· · · · · · ·

ί.

ł

Ľ

٠.

TABLE 8 (CONTINUED) - PS2 3923+25

 \mathbf{p}

767		10		772	-		375	***	240	IEC	613	-	225				212						-	164	634	-	5 3	523	668	133	16E		, C ()	184	510	6.9	5.0	303	1.41	372	773	5		5.0	5		-
. 58 .							1971 1971 1971 1971		EET.		. 533	•••	• 516	• 506	• 593	5.1		EFZ.							.55	.50	•159	• 335	516.	•17 <u>5</u>					5			521.	C		5.5	613 ·		524		Nº C .	();;;
2913				6931	1640	4004	65 i E	0.883	2062	5942	145E	1713	4985	8755	5610	6172	5666	9534					19561	1513	2013	2313	1925	7329	197	61 F3	4646			111	1120	543	-57	53	772	7556	2856			1126			
0		••	4 5 0	n	1	0 0	0	0	60	е С	0	0	0	0	•	9 0	9 0	0 0		0 A	• •	•	•	*	- M		2 7	2	ч -1	-		ri u	-	- 0		~	N	N	~		in (N	10 N (10 U N 1	0 4 N (1
68.				5C1-	÷105	-177	M 10 -	C9-	-31	÷5=	C2.	***	şe.	-12	2	10 1 1	-21					• m	, n	32	•	12	•	*	5 2							12.	•	•	~		•	ب	•	2	<u>, , , , , , , , , , , , , , , , , , , </u>	n 4 	
.236381				225775	\$E14Ca+	. 235887	8264024	· 201603	1.252917	1.200972	1.137239	1+23365	1.231822	680961 - I	1.159137	E92225+1	9120021	eEE2C2+I						669804+	1205066	+60208+	EC54C2 .	4044Cer	+55+65+	ESEEC2+				Eccare.	1029561	111905.	• 255595	• 255733	• 235536	• 25337	1EEC: .		•19352¥	• 272533	264202.	- 500250	
4					-	4	*	4	•, •,	-	E		-	*	4	-	*	*					-	-		-	-	*		1			* *			*	*	*	-	*	*	*	*	*		10 A 18 A	•
in	•					C 1			e e	-	2		•	1 49	Ē	Ē		2		0	" "		• •		11	11	-	*	15	'n	-		ົລ	9 60	1	3 7	11	"	*	*	•		15	Ē,	# 1	n,	4
982•317325		2010-00-00 200-597596	642+320055	961707-221	522-553226	762-572027	522+533266	242+715003	383+ 335245	323+47 ₀₀ 93	123+225310	223+ 35 ₂₀₁ 3	163.437247	243+271617	\$83-616120	523+722943	303+5+5275	563+526153				LB3-258570	143-557659	01954950	544+717591	544+532015	*****73627	044-641023	P34+241634	544+471207	004-264997		362604.404	964954-12	84-551651	94+433573	44+327368	664.55391D	84+442724	584+4J345	79644.243997	284+615553	<u>24+73+832</u>	14.277945	304+3'+126	554.036470	
12			. m		2	80	m S	~	5 83	5 36	*	72	5	19	4	99	2	55						-	E		Š.	5 27	5 27	ŝ				ī		Ē	-	-		Ě	5	at i ar	ŝ	E 9 - 5	1	ŝ	
142						147	147	147	13.1	137	1576	1340	1540	154	154	134	156	139					157	157	1.5	154	159	1590	1536	171			177	7.83	17.	1713	19-61	15.1	19.1				157				
* 6.						•	* E1	• 6	• 20	5	•	32 •	•	• 55		12 •	*		•	• •) •		• • •		•	30 30	•	* *	•	* : • •	••		•	* 90	ي 2	•		•	م	* 2'	:	•	•	• •	
67 9 99•3374(51729+5953	55449+5196	++312+2430C	48513+7550	57291 • 53314	20368+64062	32523+8974	36244 • 3525	39644+6737	34844.33335(37164+3652;	24813+7311	29774.5394;	27971 . 9328	30611+9348	27597.5712	9-897•24592			5576.2691	9576.0474	17515-9423	55683-52156	64618-5574	68398+3239	68812+3921	78712+37111	3#192+30593	6 582 • 3785		54886-2009	19-9-1284	61233-25661	64413+51245	32992•19389 <e< th=""><th>39382+7132;</th><th>17723+++771</th><th>21 423 . 1 41 82</th><th>41183+2873</th><th>12555-C95F4</th><th>32333.66<u>1</u>47</th><th>36412+72517</th><th>19464+C6646</th><th>2065-29205</th><th></th></e<>	39382+7132;	17723+++771	21 423 . 1 41 82	41183+2873	12555-C95F4	32333.66 <u>1</u> 47	36412+72517	19464+C6646	2065-29205	
0	•	> c	0	0	0	0	C	n	0	C	0	0	0	0	0	n	0	n	0	00	5 C	o c	0	0	0	0	Ô	0	0	0	0	0 0	20) c	0	0	0	0	n	0	n	0	n	c	n i	C	
9 9			22	95.	92.		-	-21	65.	*E-	-72	2	i	ŗ	96.		Ì					.7.		E.C. a	751.		24.			•					-21	86-	69.	-28	8 /-	•10B			111.	123	-117		
+0+512299			\$0-863652	42+763665	40-932697	10559+04	40-9-5975	41-032667	76145c+14	1.037,38	\$016EC+1+	\$E6622.14	41.227499	874CCE -14	196006.14	41+352652	1-349327	\$1 • 355755	4:-355776			151214-14	41.481122	41.479656	41-555485	*1.546163	41+550005	C+E543+14	41+579092	+1.627172	41.695659		++6C22-1+	41 c 83 3474	840E+1+1+	167148-14	41-994583	41+99R035	41-99f915	\$2+065733	5.547C.54	12.073371	42.23389	42+234111	266102+24	//6662+24	
272				10	31	8		-	1	8	5	#	13	20	4	125	5					ם או ל וו			2	-	!	5	16				- 31 	36	8		•	•	5	1	•	n (R i -	C 1	£. (
68447+853915,2		477444939134	52-161-235131	5579894117	59449C-14654	48541-246238	57221-192647	146256 - 14588	1924C1+12626	36241+223425	39541 . 424477	24361-563192	96731-275353	25421+549916	29391-683718	27721-575394	30341+623736	27341-531839	77555.483579	17331.579570	21551-51-55 21581-58-155	5431-655479	002664-1046	12641-925948	96-51.0 15C08	54451-557453	58441-971954	12cC28.16265	79192+153935	82661 - 805734	52081+909999	7666666799998	55112-197757	245651.05164	\$1481+9379 ₂ 9	6+561+93 <u>5</u> 92	32742+2333555	38142+292126	46782+273273	265614+24013	39759+534939	121164-2464	32442+311539	35322+355454	04624+2646	126//C+26EC.	
126				248	999	595	863	606	606	606	6 06	586	586	1399	1110	1030	CECI	2601					6461	6401	2011	1:08	1108	1135	1.35	1135	1163		1192	1	1219	1219	1279	1279	1279	SCE1	SCE I	1305	1 937	LEE I	1.861 1.661		

74.

n an an an and a second and a

444546 284 (CANTINUE), 854 2423+26

000

5

-

2

•	91944-239	502+165E9 (58552+125	1221-20EC9 0					ZEZ-EC/A+ C		E18.64628 0	51333.513	2245-129	905+585+ 0	79555-592	81755-514	E41+56801 0	545+215E9 C	65292+325	EEC+CCLE9 C	65465-232	710-12-17	77357.355	72545.723	7116-61646	1212-25166 0	67151-52-55	245+EE89 (66394.513	69134-367	842.42449 0	140-40299 0	59524+945	61355.225	362444585 (6054449509				22777.55	EL1-76145	52/54.393	54553+231	75337.1229	19551+3993	9172.951	549494+545	- 50574+455	725354.42
	2	1		-		 	•					2	~	5	66.	-12	42.		C.2.	5	21.	-	~		~	611	.0)) 	, ,		2	-		m d	•	5					۰ ۲	1		35	413 S	41 A	# ·		• 12
	1EC2C2.94 8	860104+44 9	5 45.211375	1 45.199797				5 46+231682	REIZCING Z	F15125+24 7	1 47+23522J	5 47.13252	151005+14 1	6 47.232675	116102021 1	5 47.23487	5 7.135666	1 195653	2 4. 197155	7 196653	2 4/198385	5 47+234462	9 47.233964		6 47.233667	2 47+234355	Teresta 8	7 47+232664	195000-14 5	1 47+195533	0 47.199235	0 474197760	5 47+195738	4 47.197139	E66561+2+ C	2 k7.193599	1 47+202555			6464C4.64 5	5 45.274752	5 49.233392	C644C2+64 <	554900 1 C	3 49.192551	1611914	5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	5 6472337	242661+44 2
	21 22 + + 1 375 15	1 84,004+9065	3575.474169 1	7 26215. 36219 7	1000 1000 1000 1000 1000 1000 1000 100			8766+211433 20	CATIE CALL	16695125555 4	12477+234254 ?	11327+319790	2217+656230	+357+535933 :	19787-6541.55 29	11857+67+252 3	1 175/07+7511'	14227•657 <u>3</u> 91 .	5737+249258	54137.433691 3	55997+555422 2	16727.732636	8457+676632 2	13147-234446 23	1917-198586 16	68.7.459616 5	7607.443577	9357++56991	6347+531277 1	8387.273922 3	4547+644553 3	16347.472557 2	19547-247694	i1317•355928	5 57911511972 2	• 0307• 293323 •	5527+395247 6			1479.49 132	945416.86864	32874.325234 2	14755+350120 5	19411+227272 B	9.55+55544	9645.237658	13942+34891 1 1	529229252	0 BCSCC++BbdE/
	1386 . 2545 C	9 9 9 9 5 4 9 9 9 9 9 9 9 9 9 9 9 9 9 9	912					6 - 2751	4227 × 2743 4	4 55 · · · · 2243	5 i.c. 963	'945, , 2996 <u>9</u>	53886 . 2942	11036 . 2942	53621 . 23-5 7	1675 . P345 9	13572 . 2319 7	24779 . 2347 6	2414 . 2347 6	55243 . 2942 6	<u> 3547 e 2952 6</u>	14217 . 2975 7	73857 . 2975 7	5347 . 2349 J	5481 . 2949 7	9256 . 3772 6	A THE . 66 171	3692 . 3317 5	16342 . 3345 5	2 24CE - 260C	76664 + 3349 5	12055 + 309 5	7053 . 3744 5	15324 . 3 ₃ 94 6	1529 + 2175 5	5 Srlf 85453	1 ccle + 3195 /			7145 . 48444	73667 . 3217 3	2331 + 3245	38511 • 3245 •	550P6 • 33P1 7	1942 . 3329	J3647 + 3329	1944 - 3344 5	14333 • 3344 -	79631 + 33 ⁴⁷ 7
	2 50441417	21+26924 2	2 21355.20		1 44677.46			0 16595	1 17565 C	0 12795+52	24+24C4 C	C 9382.27	2 7484.25	2 A654.14	0 +0.4.53	1395.44	51445 C	0 34156+12	0 73601.12	2 76241.15	0 70161.99	0 71662.17	0 66564.67	2 51015.75	0 61575.65	0 61416.69	0 55342.67	2+E-285 0	+C+E1EC4 C	0 22953+61	3 25527.67	3 26137.59	3 24752.61	3 37457+15	3 34362+13	3 27319+52	C#+#26C2 0		3 85114 439	3 55703-94	3 1745.27	3 4365.51	0 74157+73	50-70921-95	0 72512.71	C.Edi+7 C	2 74165+74	0 77 35+34	51-91C H 9 C
,		-	7				22.	18.	B•	0	36 - 36	÷5- 6-	5 -17		·2. C	E2- 19	1 -18	12. E	13 -10	-10	11 13	•	6 14	46 0		6 27	5		5 18	17 16	F. 6:	5	9 9	10	-1-	CS+ E7				52 -17	5 -12	-16	53 -16	53 -17	C es	11- CI	4 ·	61 E	15 8
	4602+44	10191.04	19991-54	5.0.3				+3-2-64	42+5024 42+506	45.2-65	+5+205C	•3•2 ¹ 326	+5+2-1281	16104-54	13.21280	15.2716	15+27260	45.2-14	45.2.28	+105+54	\$5+2009	45+1999	45-1989	+5+2-33	1999	45-1956	+5+210+C	45-1975	42-1942	+5+1959	46.2344	46+2345	\$6+204B	96-2029	14002+9t	\$6102+94	46-203	F 1 6 1 • 0 •		\$5.2359	46-2341	16.2337	42.0.244	46+2033	45.235	+5+231	+0-2-64	+5+1977	1605+54
	•	16					1	•			401	•12	9 9	52	20	11	84	159	36	10	9			•	-	-		32	~	: 6 . 110	•	9 1	18	30	A. 11	60 #	*		9 - 6	~	•	40 •	'n	C.		đ.	n n
-	52424-24935	4cE4C5.5EcC2	7494.62277.					16195+357597	1.4053.4.7517	12725+717593	/155+651473	064CC1+52č6 .	7755+326556	8985+2+29334	+3.377513	1945+453479	120303023131	C19466-24446 .	. 74-155-395413	76725.554237	10545+335332	72-45	50759+6:5997	53-95. 12-1677	61755+3735+3	48125-26465	35185.425275	555'5-697528	2119C2+SC+61 .	22345-525941	C46169-14655 1	1 25451+534535	26255+364445	1 35955.723154	115659-16516 .	· 23=95+594439	20333+635533	VINC+5+6C67/ 2	1 95429-727411	+75215+21597+	+295-3+952+	1 4735+734958	76555+40050	78425+535494	710464-9CEE1 1	+Ec1C2-552+2 1	1 75535+435511	1 77535+54545333	55445+5+54132
•	1 91 c	235-	2383					6E 1 a	56.25	6418	177¢	4474	1 614	a198	9222	9000	6tae	5239	1.000	2265	1672	462č	lcEč	2336	SEES	9562	1950	£354	2002	50.45	CENS	CENE	CEŧë	545	5.65	225e			6928	2551	255g	2549	C15ë	2125	553	529č	5896	5.000 1000	555

5 N N N N N

5

n /

2

 26

19 C

1.1

<u>ب</u> ب	•	8-197649	-		75145+3966	154 .	645E	2719	9-23353	•	64		•	5	19.62
-	-	9-193133	Cal	4	64377+J235	570 .	9349	2751	9.23155	7 11		1235754	-	5	5.9.926
ě	•	8+193135	5	4	68504+3755	. 940	3533	5135	7.65.87	2 76	64	197256	54.		
Ç	•	8.200071	04	4	55325+1351		19547	646	5.71683	95	Č.	205665			574.39
n	-	197327	•	4	59369.4144		3573	7843	1.65820	1 24	6 4	1975257		. 75	2.6.640
	*	3+195263	•	4	56121 + 5720	. 203	3710	6952	9.4.72831	69		24145			12.0.00
Ŕ,	•	5-191932	•	•	59325+9745	• • • •	0110	7111	1.37599	3 37	ŝ	164452			6.16.1
•	•	5+274003	202	5	42943+4572	. 25	3741	7462	5.231145	3 1	64	195725	;		5.9.5
Ē	4	9-155781		5	14341+522	76A .	3751	7626	5.4.35931	-		9792	-	12	111111111111111111111111111111111111111
۳ñ	• c	5.23376	17	in	40512.733E	. 642	3751	7692	7.426191	52	Ē	196700	•		
ē	•	19261.5		4	55714.5334	• • • •	CP/E	5417	1.54261	5		1199375	-	E9 E1	928.75
ě	•	6-157957	-	*	56511.2355		1710	545.	3•5952 <u>3</u> 1		÷.	569614			232.5
_	•	9+199755	11	n	\$551.5386	• 614	CF	E 4 9 9	5+5+654		6.4	199352	-	99 E I	
í E	•	9.234681	-	-	3721 ⁵ •7693		35.38	5533	7.55649			1250041	11	- -	10.010
-	-	9-20414	-	in	33606+8633	. 661	8-6E	5000	5.47541	203	E.	Scrord of		-	11.13
-	•	9.25854			27338.6336				3.24945						

REFERENCES

- Hewish, A., Beil, S. J., Pilkington, J. D. H., Scott, P. F., and Collins, R. A., "Observation of a Rapidly Pulsating Radio Source," Nature, Vol. 217, 1968, p. 709.
- Pilkington, J. D. H., Hewish, A., Bell, S. J., and Cole, T. W., "Observations of Some Further Pulsed Radio Sources," <u>Nature</u>, Vol. 218, 1968, p. 125.
- Gold. T., "Rotating Neutron Stars as the Origin of the Pulsa.ing Radio Sources," Nature, Vol. 218, 1968, p. 731.
- 4. Moffet, A. T., and Ekers, R. D., "Detection of the Pulsed Radio Source CP 1919 at 13-cm Waverength," <u>Nature</u>, Vol. 218, 1968,
 p. 227.
- Bracewell, R. N., The Fourier Transform and Its Applications, McGraw-Hill, New York, 1965, p. 339.
- Foster, C. F., "Automated Pulsar Receiver," <u>DSN Progress</u> <u>Report 42-20</u>, Jet Propulsion Laboratory, Pasadena, California, April 15, 1974.
- Slekys, A., <u>A New Pulsar Timer</u>, JPL Technica! Report 32-1526, Vol. XIII, Jet Propulsion Laboratory, Pasadena, California, Feb. 15, 1973.
- Brokl, S. S., "Automated Pulsar Data Collector," <u>DSN Progress</u> Report 42-25, Jet Propulsion Laboratory, Pasadena, California, Feb. 15, 1975.
- 9. Reichley, P. E., Downs, G. S., and Morris, G. A., "Time-Of-Arrival Observations of Eleven Pulsars," Ap. J., Vol. 159, 1970, p. L35.

- Cooley, J. W., Lewis, P. A. W., and Welch, P. D., "The Fourier Transform Algorithm: Programming Considerations in the Calculation of Sine, Cosine and Laplace Transform," J. Sound Vib., Vol. 12, 1970, p. 315.
- Brigham, E. O., <u>The Fast Fourier Transform</u>, Prentice-Hall, Englewood Cliffs, New Jersey, 1974.
- <u>NBS Time and Frequency Bulletin</u>, U. S. Dept. of Commerce, National Bureau of Standards, Boulder, Colorado, 1968 to 1979.
- Daily Phase Values and Time Differences, Series 4, U. S. Naval Observatory, Washington, D. C., 1968 to 1977.
- <u>Times Services Bulletin</u>, Series 11, No. 220, U. S. Naval Observatory, Washington, D. C.
- Preliminary Times and Coordinates of the Pole, Series 7, No. 210,
 U. S. Naval Observatory, Washington, D. C.
- Melbourne, W. G., Mulholland, J. D., Sjorgren, W. L., and Sturms,
 F. M., <u>Constants and Related Information for Astrodynamic Cal-</u> <u>culations, 1968</u>, JPL Technical Report 32-1306, Jet Propulsion Laboratory, Pasadena, California, July 15, 1968.
- Sturms, F. M., Jr., Polynomial Expressions for Planetary Equators and Orbit Elements with Respect to the Mean 1950.0 Coordinate System, JPL Technical Report 32-1508, Jet Propulsion Laboratory, Pasadena, California, January 15, 1971.
- 18. Moyer, T.D., Transformation from Proper Time on Earth to Coordinate Time in Solar System Barycentric Space-Time Frame of <u>Reference</u>, JPL Technical Memorandum 33-786, Jet Propulsion Laboratory, Pasadena, California, Dec. 1, 1976.

 Koble, H. M., Pease, G. E., and Yip, K. W., "LS44-An Improved Deep Space Network Station Location Set for Viking Navigation," in Deep Space Network Progress Report 42-35, Jet Propulsion Laboratory, Pasadena, California, Oct. 15, 1976.

- White, R. J., Rosenberg, A. D., Fisher, P. S., Harris, R. A., and Newhall, N. S., <u>SPACE Single Precision Cowell Trajector Program</u>, JPI Technical Memorandum, 33-198, Jet Propulsion Laboratory, Pasadena, California, Jan. 15, 1965.
- Holdridge, D., Space Trajectories Program for the IBM 7090
 Conputer, JPL Technical Report 32-223, Rev. 1, Jet Propulsion Laboratory, Pasadena, California, September 1, 1962.
- 22. Taylor, J. H., and Manchester, R. N., "Observed Properties of 147 Pulsars," Astron. J., Vol. 80, 1975, p. 794.

...

.