

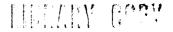
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SPAR DATA SET CONTENTS

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SUMMARY

The SPAR structural analysis system consists of a collection of processors for performing finite-element analysis. The data generated by each of these processors are stored in a data base library as two-dimensional tables or matrices called data sets. These data sets are identified by four-word names which are listed in the table of contents for the data base library.

This report documents the contents of the SPAR data sets. The creating SPAR processor, number of rows and columns, and definitions of each of the data items are documented for each data set. An example problem is included, with SPAR input and resulting table of contents. This information can be used to create new SPAR processors or to interface SPAR with another system.

INTRODUCTION

The SPAR computer software system is a collection of processors for performing finite-element structural analysis. The data generated by each processor is stored on a data base complex for use by subsequent processors. This data may be read by the user through SPAR data processors (ref. 1) or FORTRAN data handling utilities (ref. 2).

This report documents the contents of many of the SPAR data sets stored in the data base complex. An example problem for creating the data sets documented in this report is included, with the SPAR input and resulting table of contents listed. This data was collected in the course of writing an interface from SPAR to another data base complex, and represents the formalizing of a set of notes passed on by several users. This information can be used (1) to understand more clearly and to use more productively the existing processors in the SPAR system, (2) to develop new SPAR processors, or (3) to interface SPAR with another software system.

SPAR DATA SET CONTENTS

The SPAR software system, created by W. D. Whetstone, is composed of processors for performing finite-element structural analysis. The data from a SPAR run is saved in a data file organized as a "library." Each library file is composed of a number of datasets and a table of contents. These library files have names SPARLA through SPARLZ for Control Data; SPAR-A through SPAR-Z for Univac; or SPLA through SPLZ for PRIME or VAX; corresponding

to SPAR library numbers 1 through 26. The data is automatically put in library 1 by the processors. This file may be renamed to another library name and may still be read using DCU or the data handling utilities.

The data are stored in 2-dimensional tables or matrices dimensioned (NI, NJ) called blocks. Each data set contains one or more blocks with NI rows and NJ columns. Following are lists of the contents of many SPAR data sets. The creating SPAR processor, number of rows and columns, and the definition of each of the data items are listed for each data set.

JDF1 BTAB 1 8

```
Created from TAB processor START card.
NJ = 1
NI = 18
```

Contents:

- 1. Total number of joints.
- 2. Number of joint degrees of freedom, both translations and rotations, constrained by START card.
- Number of joint translational degrees of freedom not constrained.

```
4.
        A list of unconstrained joint degrees of freedom, filled in
 5.
        consecutively from position 4; unused values are zero.
 6.
        Example for d.o.f. 1, 2, and 6 unconstrained:
 7.
             1,2,6,0,0,0
 8.
 9.
10.
        A list specifying the order of each unconstrained degree of
11.
12.
        freedom; zero if not active.
        Example for d.o.f. 1, 2, and 6 unconstrained:
13.
14.
             1,2,0,0,0,3
15.
16.
17.
        Not used.
```

JREF BTAB 2 6

18.

```
Created from JREF in processor TAB.
NJ = Total number of joints
NI = 1
```

Contents:

Contains the Joint Reference Frame number for each joint, corresponding to the row number of dataset ALTR BTAB 2 4 which contains the definition of each Joint Reference Frame.

ALTR BTAB 2 4

Created from ALTREF in processor TAB.

NJ = Number of alternate reference frames
NI = 12

Contents of each entry:

```
1.
       a<sub>11</sub>
 2.
       a<sub>21</sub>
 3.
       a31
 4.
       a<sub>12</sub>
                   Components of a
 5.
       a22
                   coordinate transformation matrix
 6.
       a<sub>32</sub>
 7.
       a<sub>13</sub>
 8.
       a<sub>23</sub>
       азз
Х<sub>о</sub>
10.
                   Location of origin of
11.
                   alternate reference frame
12.
                   given in global coordinates
```

Formula:

$$\begin{bmatrix} Xg \\ Yg \\ Zg \end{bmatrix} = \begin{bmatrix} a_{11} & a_{12} & a_{13} \\ a_{21} & a_{22} & a_{23} \\ a_{31} & a_{32} & a_{33} \end{bmatrix} \begin{bmatrix} Xa \\ Ya \\ Za \end{bmatrix} + \begin{bmatrix} X_o \\ Y_o \\ Z_o \end{bmatrix}$$

coordinates in global reference frame

coordinates in alternate reference frame

NDAL 0 0

Created from TITLE card in processor TAB.

Contains title in text.

TEXT BTAB 2 2

Created from TEXT card(s) in processor TAB.

Contains data in text.

JLOC BTAB 2 5

Created from JLOC in processor ${\tt TAB}$.

NJ = Number of joints

NI = 3

Contents:

MREF BTAB 2 7

Created from MREF in processor TAB.
NJ = Number of beam orientation entries

NI = 5

Contents of each entry:

(See reference 1 description of MREF input)

Format 1 (Default)

- 1. Beam axis NB
- Global axis NG
- 3. 1 if cosine between NB and NG is positive, -1 if negative
- 4. Cosine of angle between NB and NG
- 5. 1. indicating format = 1

Format 2

- 1. X₁
- x_2
- 3. x_3^2
- 4. Il axis orientation
- 5. -1. indicating format = 2

MATC BTAB 2 2

Created from MATC in processor TAB.

NJ = Number of material types

NI = 10

Contents of each entry:

- 1. E = Modulus of elasticity
- 2. v = Poisson's Ratio
- 3. G = E/(2(1+v))
- 4. ρ = Weight per unit volume
- 5. α_1 = Thermal expansion coefficient, direction \overline{x}
- 6. α_2^2 = Thermal expansion coefficient, direction y
- 7. Θ^2 = Angle between element reference frame (x,y) and (x,y)
- 8.
- 9. Not used
- 10.

BA BTAB 2 9

Created from E21 section properties in processor TAB.

NJ = Number of entries

NI = 31

Contents of each entry:

(See reference 1 description of DSY input of E21 section properties)

1. 2. 3. 4. 5. 6. 7. 8. 9. 10. 11. 12. 13. 14. 15.	Element type indicator Not used I C 1 C 2 C 2 a f f 1 z 1 z 0 4 q q q q q q q q q q q q	17. 18. 19. 20. 21. 22. 23. 24. 25. 26. 27. 28. 29. 30. 31.	Number of points at which stresses are to be calculated y11 y12 y21 y22 y31 y32 y41 y42 b1 t1 b2 t2 b3 t2
то.	^q 3)T.	² 3

BC BTAB 2 11

Created from E23 section properties in processor TAB.

NJ = Number of entries

NI = 6

Contents of each entry:

- 1. Cross-sectional area of axial element
- 2. Cross-sectional area of axial element

3.

· \ Not used.

٥,

SA BTAB 2 13

Created from shell section properties in processor TAB.

NJ = Number of entries

Contents vary according to section type:

For MEMBRANE, PLATE, ISOTROPIC or UNCOUPLED section types NI = 43

```
Contents of each entry:
 1. Number indicating section type:
                                                             16-25.
                                                                            Not used.
               1 = MEMBRANE
                                                             26.
                                                                     f<sub>11</sub>
                                                                    f_{21}^{-}
                2 = PLATE
                                                             27.
                                                                    f<sub>31</sub>
f<sub>41</sub>
                3 = ISOTROPIC or UNCOUPLED
                                                             28.
       Pointer to entry of NMAT
                                                             29.
        containing material constants
                                                                    f 51
f 61
f 12
f 22
                                                             30.
 3.
        Structural weight/area
                                                             31.
       d
d
11
d
12
d
22
 4.
                                                             32.
 5.
                                                             33.
 6.
                                                                    f32
f42
f52
f62
f13
f23
f33
                                                             34.
                                                                                    stress
       d<sub>13</sub>
 7.
                                                             35.
                                                                                    coefficients
       d_{23}
 8.
                                                             36.
       d<sub>33</sub>
 9.
                       flexibility
                                                             37.
       \widetilde{d_{44}}
10.
                       coefficients
                                                             38.
       d45
11.
                                                             39.
12.
                                                             40.
       d<sup>55</sup>
13.
                                                             41.
                                                                    f<sub>43</sub>
f<sub>53</sub>
f<sub>63</sub>
          46
       d<sub>56</sub>
14.
                                                             42.
       d<sub>66</sub>
15.
                                                             43.
        For COUPLED section types:
        NI = 43
Contents of each entry:
 1. Number indicating section type:
               4 = COUPLED
 2. Pointer to entry of NMAT
        containing material constants
       Structural weight/area
                                                             25. Number of layers
                                                             26.
 4.
       d_{11}
                                                                    f<sub>11</sub>
f<sub>21</sub>
f<sub>31</sub>
f<sub>41</sub>
f<sub>51</sub>
       d<sub>12</sub>
                                                             27.
 5.
       \overline{d}_{22}
 6.
                                                             28.
       d_{13}
 7.
                                                             29.
 8.
                                                             30.
       d<sub>33</sub>
 9.
                                                             31.
                                                                    f<sup>61</sup>
       d<sub>14</sub>
10.
                                                             32.
                                                                    f 12
f 22
f 32
f 42
       d<sub>24</sub>
                                                             33.
11.
       d34
d34
12.
                                                             34.
                                                                                    stress
                                                             35.
13.
                                                                                    coefficients
                                                                    £42
                       flexibility
       d15
14.
                                                             36.
                                                                    f52
                       coefficients
                                                             37.
15.
                                                                    \tilde{f}^{62}
                                                             38.
16.
       d35
                                                                    -13
17.
                                                             39.
       \tilde{d}^{45}
                                                                    f<sup>23</sup>
18.
                                                             40.
                                                                    £33
       d55
d16
d26
d36
19.
                                                             41.
                                                                    £43
                                                             42.
20.
                                                                     f<sub>63</sub>
21.
                                                             43.
22.
        ď
       d<sup>46</sup>
23.
24.
         66
```

For LAMINATE section types: NI = 25 + (18 times number of layers)

```
Contents of each entry:
    Number indicating section type:
                                              25.
                                                   Number of layers
                                                                            44-61.
           5 = LAMINATE
                                              26.
                                                   <sup>g</sup>11
                                                                                     for 2nd
     Pointer to entry of NMAT
      containing material constants
                                                                                     layer
                                              27.
                                                                            62-?
 3.
      Structural weight/area
                                                                                     Eighteen
                                              28.
                                                                                     additional
 4.
     d_{11}
                                                                                     values for
 5.
                                              29.
                                                                                     each
 6.
                                              30.
                                                   g<sub>22</sub>
                                                                                     successive
 7.
                                                                                     layer.
 8.
                                              31.
 9.
                                              32.
10.
11.
                                              33.
12.
                                              34.
13.
               flexibility
                                                            layer stress
14.
                                              35.
               coefficients
                                                            recovery
15.
                                                            coefficients
                                              36.
16.
17.
                                              37.
18.
                                              38.
19.
20.
                                              39.
21.
                                              40.
22.
23.
                                              41.
24.
                                              42.
                                              43.
```

COil ncon O

ncon = Constraint case
Created from CON in processor TAB
NJ = Number of joints
NI = 1

Contents:

Each entry contains an integer representing the Joint Reference Frame number and constrained components for that joint. The bit pattern of each integer contains a 1 for constrained components, zero otherwise, stored in reverse order (6 to 1) with Joint Reference Frame number leading. For example:

A joint with components 1,2,3, and 5 zeroed out and JREF = 11 would have the integer 45333 stored:

bit pattern
$$\rightarrow$$
 0. . .01011 00 01 00 01 01 01 JREF number $6 \quad 5 \quad 4 \quad 3 \quad 2 \quad 1$ joint motion components

```
Component 1 (constrained)
                                1 \times
Component 2 (constrained)
                               1 ×
                                       4 =
                                                  4
                            ==
Component 3 (constrained)
                                1 ×
                                      16 =
                                                 16
Component 4 (unconstrained) =
                               0 ×
                                      64 =
                                                  0
                           = 1 \times 256 =
Component 5 (constrained)
                                                256
Component 6 (unconstrained) = 0 \times 1024 =
                                                  0
                            = 11 \times 4096 = +45056
JREF number = 11
Integer stored for this joint
```

QJJT BTAB 2 9

Created in processor TAB.

NJ = Number of Joints

NI = 9

Contents of each entry:

- 1. a₁₁
- 2. ^a21
- 3. a₃₁
- a_ 22 5.
- 6.
- 7.

Formula:

Each entry contains a 3×3 matrix to convert Alternate Reference Frame to Global Reference Frame for that joint.

$$\begin{bmatrix} Xg \\ Yg \\ Zg \end{bmatrix} = \begin{bmatrix} a_{11} & a_{12} & a_{13} \\ a_{21} & a_{22} & a_{23} \\ a_{31} & a_{32} & a_{32} \end{bmatrix} \begin{bmatrix} Xa \\ Ya \\ Za \end{bmatrix}$$

coordinates in global reference frame

coordinates in alternate reference frame

DEF Exx y z

Exx = Element name

= Type number (E21 = 1 through E44 = 12)

= Number of joints/element

Created from element definitions in processor ELD.

NJ = Number of elements of this type

NI For 2-node elements = 18 columns

3-node elements = 15 columns

4-node elements = 16 columns

3.50

Contents of each entry:

- 1. Element number
- 2. Group number
- 3. Element number within group
- 4. Not used
- 5. N3 of corresponding dataset xx BTAB N3 N4
- 6. N4 where xx = BA, BB, SA...
- 7. Index of MATC containing material constants
- 8. Index section property dataset containing section properties.
- 9. Index of non-structural weight dataset (NSW)
- 10. Index of rigid link offset dataset (BRL)
- 11. Index of beam orientation dataset (MREF)
- 12. Not used
- 13. Joint #1
- 14. Joint #2
- 15. Joint #3
- 16. Joint #4
- 17. Not used
- 18. Not used

GD Exx y z

Exx = Element name

y = Type number (E21 = 1 through E44 = 12)

z = Number of joints/element

Created from element definitions in processor ELD.

NJ = Number of groups

NI = 2

Contents of each entry:

- 1. Total number of elements within group.
- 2. Cumulative total of last element in each group for more than one group.

GTIT Exx y z

Exx = Element name

y = Type number (E21 = 1 through E44 = 12)

z = Number of joints/element

Created from element definitions in processor ELD.

NJ = Number of groups

NI = 15

Contents of each entry:

15 words of title for each group.

Default is blanks.

DIR Exx y z

Exx = Element name

= Type number (E21 = 1 through E44 = 12)

= Number of joints/element

Created in processor ELD.

NJ = 1

NI = 20

Contents:

- 1. Number of nodes
- 2. Type number
- 3. Number of elements of this type4. N4 in "xx BTAB N3 N4" where xx is BA,BC,SA...
- 5. Size of Exx EFIL for this element
- 6-20. Directory information for element data.

NS 0 0

Created in processor ELD.

NJ = Number of element types present

NI = 15

Contents:

Each entry contains directory information for corresponding element data.

ELTS NAME 0 0

Created in processor ELD.

NJ = 1

NI = Number of element types

Contains alphanumeric element name of each element used in the structure.

ELTS NNOD 0 0

Created in processor ELD.

NJ = Number of element types

NI = 1

Contains the number of nodes in each element type.

ELTS ISCT 0 0

Created in processor ELD.

NJ = Number of element types
NI = 1

Contains N4 of "xx BTAB N3 N4" where xx = BA, BC, SA...

KMAP y z

Created in processor TOPO.

This information is stored in blocks with the block length determined by a RESET control in the TOPO processor. Default block length is 896 words.

Contents:

Used by K, M, and KG to guide assembly of stiffness and mass matrices in the SPAR standard sparse-matrix format.

AMAP y z

Created in processor TOPO.

This information is stored in blocks with the block length determined by a RESET control in the TOPO processor. Default block length is 1792 words.

Contents:

Used by INV in factoring system matrices, such as the reduction of the stiffness matrix.

Exx EFIL y z

Exx = Element name

y = Type number (E 21 = 1 through E 44 = 12)

z = Number of joints/element

Created in processor E.

NJ = Number of elements of this type

Contents:

Each entry contains alphanumeric information packet with the following categories:

- 1. Integer information, corrected joint numbers, tables
- 2. Material constants
- 3. Geometrical details
- 4. Section properties
- 5. Intrinsic stiffness matrix
- 6. Stress recovery influence matrix
- 7. Internal stress resultants

DEM DIAG 0 0

Created in processor E.

NJ = Number of joints

NI = 6 minus number of joint motion components constrained on START card

Contains system mass matrix in diagonal form.

K SPAR ncon 0

ncon = constraint case

Created in processor K.

This information is stored in blocks with the block length determined by a RESET control in the K processor. Default block length is 2240 words.

Contains stiffness matrix composed of submatrices, each submatrix corresponding to the connection of one joint to another.

INV x ncon 0

x = First word of the name of the input stiffness matrix

ncon = Constraint case

Created in processor INV.

This information is stored in blocks with the block length determined by a RESET control in the INV processor. Default block length is 3584 words.

Contains factored system matrix.

CEM SPAR ncon 0

ncon = Constraint case
Created in processor M.

This information is stored in blocks with the block length determined by a RESET control in the M processor. Default block length is 2240 words.

Contains unconstrained system consistent mass matrices considering only the structural and nonstructural distributed mass associated with the elements.

CASE TITL iset

iset = Load set

Created in processor AUS.

Number of blocks = Number of load cases in this load set.

Contents:

Each block contains the title for the corresponding load case in text.

APPL FORC iset

iset = Load set

Created in processor AUS.

NJ = Number of joints

NI = 6 minus number of joint motion components constrained on START card. Number of blocks = Number of load cases in this load set.

Contents:

Each entry contains applied forces and moments on that joint in each active direction.

APPL MOTI iset

iset = Load set

Created in processor AUS.

NJ = Number of joints

NI = 6 minus number of joint motion components constrained on START card. Number of blocks = Number of load cases in this load set.

Contents:

Each entry contains applied motions on that joint in each active direction.

NODA TEMP iset

iset = Load set

Created in processor AUS part TABLE

NJ = Number of joints

NI = 1

Number of blocks = Number of load cases in this load set.

Contents:

Each block of data contains nodal temperatures for every joint in the structure. One block corresponds to one load case.

NODA PRES iset

iset = Load set

Created in processor AUS part TABLE

NJ = Number of joints

NI = 1

Number of blocks = Number of load cases in this load set.

Contents:

Each block of data contains nodal pressures for every joint in the structure. One block corresponds to one load case.

TEMP Exx iset icase

Exx = Element name

iset = Load set

icase = Load case within Load set

Created in processor AUS.

NJ = Number of elements of this type.

For 2-node elements:

Note: Not defined for E25 elements

NI = 3

Contents of each entry:

- 1. Average temperature of the element
- 2. Transverse gradient in direction 1
- 3. Transverse gradient in direction 2

For 3-node elements:

Note: Not defined for E32 elements NI = 3

Contents of each entry:

- 1. Temperature at joint 1 of element
- 2. Temperature at joint 2 of element
- 3. Temperature at joint 3 of element

For 4-node elements:

Note: Not defined for E42 elements
NI = 4

Contents of each entry:

- 1. Temperature at joint 1 of element
- 2. Temperature at joint 2 of element
- 3. Temperature at joint 3 of element
- 4. Temperature at joint 4 of element

Formula:

Total effective = Element temperature + Nodal temperature from block icase temperature at node n - at node n of dataset "NODA TEMP iset"

DISL Exx iset icase

Exx = Element name

iset = Load set

icase = Load case within load set

Created in processor AUS

NJ = Number of elements of this type

For 2-node elements:

NI = 6

Contents of each entry:

- 1. Displacement in direction 1
- 2. Displacement in direction 2
- 3. Displacement in direction 3
- 4. Rotation about axis 1
- 5. Rotation about axis 2
- 6. Rotation about axis 3

These displacements and rotations are relative to a reference frame, parallel to the element's reference frame, and embedded in the terminus.

For E31 elements:

NI = 3

Contents of each entry:

- 1. Displacement of joint 2 in direction 1
- 2. Displacement of joint 3 in direction 1
- 3. Displacement of joint 3 in direction 2

For E32 elements:

NI = 6

Contents of each entry:

- 1. Displacement of joint 2 in direction 3
- 2. Rotation of joint 2 about axis 1
- 3. Rotation of joint 2 about axis 2
- 4. Displacement of joint 3 in direction 3
- 5. Rotation of joint 3 about axis 1
- 6. Rotation of joint 3 about axis 2

For E33 elements:

NI = 9

Contents of each entry:

- 1. Displacement of joint 2 in direction 1
- 2. Displacement of joint 3 in direction 1
- 3. Displacement of joint 3 in direction 2
- 4. Displacement of joint 2 in direction 3
- 5. Rotation of joint 2 about axis 1
- 6. Rotation of joint 2 about axis 2
- 7. Displacement of joint 3 in direction 3
- 8. Rotation of joint 3 about axis 1
- 9. Rotation of joint 3 about axis 2

For E41 elements:

NI = 6

Contents of each entry:

- 1. Displacement of joint 2 in direction 1
- 2. Displacement of joint 3 in direction 1
- 3. Displacement of joint 3 in direction 2
- 4. Displacement of joint 4 in direction 1
- 5. Displacement of joint 4 in direction 2
- 6. Displacement of joint 4 in direction 3

For E42 elements:

NI = 6

Contents of each entry:

- 1. Displacement of joint 2 in direction 3
- 2. Rotation of joint 2 about axis 1
- 3. Rotation of joint 2 about axis 2
- 4. Displacement of joint 3 in direction 3

11/20

- 5. Rotation of joint 3 about axis 1
- 6. Rotation of joint 3 about axis 2
- 7. Displacement of joint 4 in direction 3
- 8. Rotation of joint 4 about axis 1
- 9. Rotation of joint 4 about axis 2

For E43 elements:

NI = 14

Contents of each entry:

- 1. Displacement of joint 2 in direction 1
- 2. Displacement of joint 3 in direction 1
- 3. Displacement of joint 3 in direction 2
- 4. Displacement of joint 4 in direction 1
- 5. Displacement of joint 4 in direction 2
- 6. Displacement of joint 2 in direction 3
- 7. Rotation of joint 2 about axis 1
- 8. Rotation of joint 2 about axis 2
- 9. Displacement of joint 3 in direction 3
- 10. Rotation of joint 3 about axis 1
- 11. Rotation of joint 3 about axis 2
- 12. Displacement of joint 4 in direction 3
- 13. Rotation of joint 4 about axis 1
- 14. Rotation of joint 4 about axis 2

For E44 elements:

NI = 6

Contents of each entry:

- 1. Displacement of joint 2 in direction 1
- 2. Displacement of joint 3 in direction 1
- 3. Displacement of joint 3 in direction 2
- 4. Displacement of joint 4 in direction 1
- 5. Displacement of joint 4 in direction 2
- 6. Displacement of joint 4 in direction 3

PRES Exx iset icase

Exx = Element name

iset = Load set

icase = Load case within Load set

Created in processor AUS

NJ = Number of elements of this type

For 3-node elements:

Note: Not defined for 2-node elements.
NI = 3

Contents of each entry:

- 1. Pressure at joint 1
- 2. Pressure at joint 2
- 3. Pressure at joint 3

For 4-node elements:

NI = 4

Contents of each entry:

- 1. Pressure at joint 1
- 2. Pressure at joint 2
- 3. Pressure at joint 3
- 4. Pressure at joint 4

STAT DISP iset ncon

iset = Load set

ncon = Constraint case

Created in processor SSOL

NJ = Number of joints

NI = 6 minus number of joint motion components constrained on START card

Contents:

Each entry contains static displacements for that joint in each active direction.

STAT REAC iset ncon

iset = Load set

ncon = Constraint case

Created in processor SSOL

NJ = Number of joints

NI = 6 minus number of joint motion components constrained on START card

Contents:

Each entry contains static reactions for that joint in each active direction.

STRS E21 iset icase

iset = Load set

icase = Load case within set

Created in processor GSF.

NJ = Number of E21 elements

NI = 52

Cont	ents of each entry;							
1.	Group number	27.	\mathbf{I}_{2}					
2.	Element number within group	28.	ας	•				
3.	Joint #1	29.	Area					,
4.	Joint #2	30.	£1					
5.	Max. combined P/A + bending (tension)	31.	$f_2^{\tilde{2}}$					
6.	Max. combined P/A + bending (compression)	32.	z ₁					
7.	P/A	33.						
8.	Transverse shear stress, S ₁	34.	$\theta^{\mathbf{z}}$ 2					•
9.	Transverse shear stress, S ₂	35.	q_1				•	
10.		36.	$\frac{q^2}{2}$			•		
11.	Shear force, end 1, direction 1	37.	q_3^2					
12.		38.		number	of	points	for	stress
13.	Axial force, end 1, direction 3	39.	у ₁₁					
14.	Moment, end 1, direction 4	40.	y_{12}^{11}					
15.	Moment, end 1, direction 5	41.	у ₂₁					
16.	Moment, end 1, direction 6	42.	У22					
17.	Shear force, end 2, direction 1	43.	у ₃₁					
18.	Shear force, end 2, direction 2	44.	У32					
19.	Axial force, end 2, direction 3	45.	y ₄₁					
20.	Moment, end 2, direction 4	46.	y ₄₂					
21.	Moment, end 2, direction 5	47.	b_1^{42}					
22.	Moment, end 2, direction 6	48.	t_1					
23.	?	49.	$^{\mathrm{b}}\overset{\cdot}{2}$			•		
24.	I_1	50.	t 2					
26.	α_1^{\pm}	51.	b_3^2					
	±	52.	t ₃					
			J					

STRS E22 iset icase

iset = Load set
icase = Load case within set
Created in processor GSF
NJ = Number of E22 elements
NI = 16

Contents of each entry:

1.	Group number	9.	Moment about axis 2 at joint 1
2.	Element number within group	10.	Moment about axis 3 at joint 1
3.	Joint #1	11.	Force in direction 1 at joint 2
4.	Joint #2	12.	Force in direction 2 at joint 2
5.	Force in direction 1 at joint 1	13.	Force in direction 3 at joint 2
6.	Force in direction 2 at joint 1	14.	Moment about axis 1 at joint 2
7.	Force in direction 3 at joint 1	15.	Moment about axis 2 at joint 2
8.	Moment about axis 1 at joint 1	16.	Moment about axis 3 at joint 2

STRS E23 iset icase

iset = Load set
icase = Load case within set
Created in processor GSF.
NJ = Number of E23 elements
NI = 6

Contents of each entry:

- 1. Group number
- 2. Element number within group
- 3. Joint #1
- 4. Joint #2
- 5. Force in element
- 6. Stress in element

STRS E24 iset icase

iset = Load set
icase = Load case within set
Created in processor GSF
NJ = Number of E24 elements
NI = 18

Contents of each entry:

- 1. Group number
- 2. Element number within group
- 3. Joint #1
- 4. Joint #2
- 5. Axial force at joint 1
- 6. Transverse shear at joint 1
- 7. Moment at joint 1
- 8. Axial force at joint 2
- 9. Transverse shear at joint 2
- 10. Moment at joint 2
- 11. Axial stress at joint 1
- 12. Shear stress at joint 1
- 13. Bending stress on upper surface at joint 1
- 14. Bending stress on lower surface at joint 1
- 15. Axial stress at joint 2
- 16. Shear stress at joint 2
- 17. Bending stress on upper surface at joint 2
- 18. Bending stress on lower surface at joint 2

STRS E25 iset icase

iset = Load set
icase = Load case within set
Created in processor GSF.
NJ = Number of E25 elements
NI = 16

Contents of each entry:

- 1. Group number
- 2. Element number within group
- 3. Joint #1
- 4. Joint #2
- 5. Force in direction 1 at joint 1
- 6. Force in direction 2 at joint 1
- 7. Force in direction 3 at joint 1
- 8. Moment about axis 1 at joint 1
- 9. Moment about axis 2 at joint 1
- 10. Moment about axis 3 at joint 1
- 11. Force in direction 1 at joint 2
- 12. Force in direction 2 at joint 2
- 13. Force in direction 3 at joint 2
- 14. Moment about axis 1 at joint 2
- 15. Moment about axis 2 at joint 2
- 16. Moment about axis 3 at joint 2

STRS E31 iset icase

iset = Load set
icase = Load case within set
Created in processor GSF.
NJ = Number of E31 elements
NI = 11

Contents of each entry:

- 1. Group number
- 2. Element number within group
- 3. Joint #1
- 4. Joint #2
- 5. Joint #3
- 6. T₁₁
- 7. T₂₂
- 8. T₁₂
- 9. Tractive force in x-direction Nx
- 10. Tractive force in y-direction Ny
- 11. Shearing force Nxy

Formulae:

 $S_x = Nx/thickness$ $S_y = Ny/thickness$ $T_{xy} = Nxy/thickness$

STRS E32 iset icase

iset = Load set
icase = Load case within set
Created in processor GSF.
NJ = Number of E32 elements
NI = 28

Contents of each entry:

- 1. Group number
- 2. Element number within group
- 3. Joint #1
- 4. Joint #2
- 5. Joint #3
- 6.
- 7. ?
- 8.
- 9. Mx Bending moment about x-axis at joint 1
- 10. My Bending moment about y-axis at joint 1
- 11. Mxy Twisting moment at joint 1
- 12. Qx Transverse shear in x-direction at joint 1
- 13. Qy Transverse shear in y-direction at joint 1
- 14. Mx Bending moment about x-axis at joint 2
- 15. My Bending moment about y-axis at joint 2
- 16. Mxy Twisting moment about joint 2
- 17. Ox Transverse shear in x-direction at joint 2
- 18. Oy Transverse shear in y-direction at joint 2
- 19. Mx Bending moment about x-axis at joint 3
- 20. My Bending moment about y-axis at joint 3
- 21. Mxy Twisting moment at joint 3
- 22. Ox Transverse shear in x-direction at joint 3
- 23. Qy Transverse shear in y-direction at joint 3
- 24. Mx Bending moment about x-axis at the center
- 25. My Bending moment about y-axis at the center
- 26. Mxy Twisting moment at the center
- 27. Ox Transverse shear in x-direction at the center
- 28. Qy Transverse shear in y-direction at the center

Formulae:

$$\begin{array}{lll} \text{Sx} &=& f_{4j}\text{M}_{x} & & f_{41} = ? & f_{51} = ? & f_{61} = ? \\ \text{Sy} &=& f_{5j}\text{M}_{y} & & f_{42} = f_{52} = -f_{62} = 6/(\text{thickness})^{2} \\ \text{Txy} &=& f_{6j}\text{M}_{xy} & & f_{43} = f_{53} = -f_{63} = -6/(\text{thickness})^{2} \end{array}$$

STRS E33 iset icase

iset = Load set
icase = Load case within set
Created in processor GSF.
NJ = Number of E33 elements
NI = 31

Contents of each entry:

- 1. Group number
- 2. Element number within group
- 3. Joint #1
- 4. Joint #2
- 5. Joint #3
- 6.
- 7。?
- 8. ?
- 9. Nx Tractive force in x-direction
- 10. Ny Tractive force in y-direction
- 11. Nxy Shearing force
- 12. Mx Bending moment about x-axis at joint 1
- 13. My Bending moment about y-axis at joint 1
- 14. Mxy Twisting moment at joint 1
- 15. Ox Transverse shear in x-direction at joint 1
- 16. Qy Transverse shear in y-direction at joint 1
- 17. Mx Bending moment about x-axis at joint 2
- 18. My Bending moment about y-axis at joint 2
- 19. Mxy Twisting moment at joint 2
- 20. Qx Transverse shear in x-direction at joint 2
- 21. Qy Transverse shear in y-direction at joint 2
- 22. Mx Bending moment about x-axis at joint 3
- 23. My Bending moment about y-axis at joint 3
- 24. Mxy Twisting moment at joint 3
- 25. Qx Transverse shear in x-direction at joint 3
- 26. Qy Transverse shear in y-direction at joint 3
- 27. Mx Bending moment about x-axis at the center
- 28. My Bending moment about y-axis at the center
- 29. Mxy Twisting moment at the center
- 30. Qx Transverse shear in x-direction at the center
- 31. Qy Transverse shear in y-direction at the center

Formulae:

$$\begin{array}{lll} \text{Sx} &=& f_{1j} \text{Nx} & + f_{4j} \text{Mx} & f_{1j} &=& 1/\text{thickness for i and j} &=& 1,2,3 \\ \text{Sy} &=& f_{2j} \text{Ny} & + f_{5j} \text{My} & f_{42} &=& f_{52} &=& -f_{62} &=& 6/(\text{thickness})^2 \\ \text{Txy} &=& f_{3j} \text{Nxy} & + f_{6j} \text{Mxy} & f_{43} &=& f_{53} &=& -f_{63} &=& -6/(\text{thickness})^2 \end{array}$$

STRS E41 iset icase

iset = Load set
icase = Load case within set
Created in processor GSF.
NJ = Number of E41 elements
NI = 23

Contents of each entry:

- 1. Group number
- 2. Element number within group
- 3. Joint #1
- 4. Joint #2
- 5. Joint #3
- 6. Joint #4
- 7. '
- 8. ?
- 9. Nx Tractive force in x-direction at joint 1
- 10. Ny Tractive force in y-direction at joint 1
- 11. Nxy Shearing force at joint 1
- 12. Nx Tractive force in x-direction at joint 2
- 13. Ny Tractive force in y-direction at joint 2
- 14. Nxy Shearing force at joint 2
- 15. Nx Tractive force in x-direction at joint 3
- 16. Ny Tractive force in y-direction at joint 3
- 17. Nxy Shearing force at joint 3
- 18. Nx Tractive force in x-direction at joint 4
- 19. Ny Tractive force in y-direction at joint 4
- 20. Nxy Shearing force at joint 4
- 21. Nx Tractive force in x-direction at the center
- 22. Ny Tractive force in y-direction at the center
- 23. Nxy Shearing force at the center

Formulae:

Sx = Nx / thickness

Sy = Ny /thickness

Txy = Nxy/thickness

STRS E42 iset icase

iset = Load set

icase = Load case within set

Created in processor GSF.

NJ = Number of E42 elements

NI = 33

Contents of each entry:

- 1. Group number
- 2. Element number within group
- 3. Joint #1
- 4. Joint #2
- 5. Joint #3
- 6. Joint #4
- 7. ?
- 8. ?

```
9. Mx Bending moment about x-axis at joint 1
```

- 10. My Bending moment about y-axis at joint 1
- 11. Mxy Twisting moment at joint 1
- 12. Qx Transverse shear in x-direction at joint 1
- 13. Qy Transverse shear in y-direction at joint 1
- 14. Mx Bending moment about x-axis at joint 2
- 15. My Bending moment about y-axis at joint 2
- 16. Mxy Twisting moment about joint 2
- 17. Qx Transverse shear in x-direction at joint 2
- 18. Qy Transverse shear in y-direction at joint 2
- 19. Mx Bending moment about x-axis at joint 3
- 20. My Bending moment about y-axis at joint 3
- 21. Mxy Twisting moment at joint 3
- 22. Qx Transverse shear in x-direction at joint 3
- 23. Qy Transverse shear in y-direction at joint 3
- 24. Mx Bending moment about x-axis at joint 4
- 25. My Bending moment about y-axis at joint 4
- 26. Mxy Twisting moment at joint 4
- 27. Qx Transverse shear in x-direction at joint 4
- 28. Qy Transverse shear in y-direction at joint 4
- 29. Mx Bending moment about x-axis at the center
- 30. My Bending moment about y-axis at the center
- 31. Mxy Twisting moment at the center
- 32. Ox Transverse shear in x-direction at the center
- 33. Oy Transverse shear in y-direction at the center

Formulae:

$$Sx = f_4j^{Mx}$$

 $Sy = f_5j^{My}$
 $Txy = f_6j^{5j}Mxy$
 $f_42 = f_52 = -f_{63} = -6/(thickness)^2$

STRS E43 iset icase

iset = Load set
icase = Load case within set
Created in processor GSF
NJ = Number of E43 elements
NI = 48

Contents of each entry:

- 1. Group number
- 2. Element number within group
- 3. Joint #1
- 4. Joint #2
- 5. Joint #3
- 6. Joint #4
- 7. ?
- 8. 3

```
Tractive force in x-direction at joint 1
10.
        Tractive force in y-direction at joint 1
11.
    Nxy Shearing force at joint 1
12.
    Nx Tractive force in x-direction at joint 2
13.
        Tractive force in y-direction at joint 2
14.
    Nxy Shearing force at joint 2
    Nx Tractive force in x-direction at joint 3
15.
16.
        Tractive force in y-direction at joint 3
17.
    Nxy Shearing force at joint 3
        Tractive force in x-direction at joint 4
18.
    Nx
19.
        Tractive force in y-direction at joint 4
20.
    Nxy Shearing force at joint 4
21.
    Nx Tractive force in x-direction at the center
22.
        Tractive force in y-direction at the center
23.
    Nxy Shearing force at the center
24.
    Mx Bending moment about x-axis at joint 1
25.
        Bending moment about y-axis at joint 1
    Mxy Twisting moment at joint 1
27.
    Qx Transverse shear in x-direction at joint 1
28.
        Transverse shear in y-direction at joint 1
29.
    Mx Bending moment about x-axis at joint 2
30.
        Bending moment about y-axis at joint 2
    Му
    Mxy Twisting moment at joint 2
31.
32.
    Ox Transverse shear in x-direction at joint 2
33.
        Transverse shear in y-direction at joint 2
34.
    Mx Bending moment about x-axis at joint 3
35.
        Bending moment about y-axis at joint 3
36.
    Mxy Twisting moment at joint 3
37.
    Qx Transverse shear in x-direction at joint 3
38.
    Qy Transverse shear in y-direction at joint 3
39.
    Mx Bending moment about x-axis at joint 4
40.
    My Bending moment about y-axis at joint 4
41.
    Mxy Twisting moment at joint 4
42.
    Qx Transverse shear in x-direction at joint 4
43.
    Qу
        Transverse shear in y-direction at joint 4
    Mx Bending moment about x-axis at the center
44.
45.
    My Bending moment about y-axis at the center
```

Mxy Twisting moment at the center

Qx Transverse shear in x-direction at the center

Transverse shear in y-direction at the center

Formulae:

Qу

46.

47.

48.

$$\begin{array}{lll} \text{Sx} &=& f_{1j} \text{Nx} &+& f_{4j} \text{Mx} \\ \text{Sy} &=& f_{2j} \text{Ny} &+& f_{5j} \text{My} \\ \text{Txy} &=& f_{3j} \text{Nxy} &+& f_{6j} \text{Mxy} \end{array} \qquad \begin{array}{ll} f_{1j} &=& 1/\text{thickness for i and j} &=& 1,2,3 \\ f_{1j}^{1j} &=& f_{1j}^{1j$$

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STRS E44 iset icase

iset = Load set
icase = Load case within set
Created in processor GSF.
NJ = Number of E44 elements
NI = 8

Contents of each entry:

- 1. Group number
- 2. Element number within group
- 3. Joint #1
- 4. Joint #2
- 5. Joint #3
- 6. Joint #4
- 7. Element thickness
- 8. Shear stress

BUCK MODE iset ncon

iset = Load set

ncon = Constraint case

Created in processor EIG.

NJ = Number of joints

NI = 6 minus number of joint motion components constrained on START card Number of blocks = Number of eigenvectors

Contents:

Each block of data contains an eigenvector corresponding to an eigenvalue stored in "BUCK EVAL." Data is stored for each joint in each active direction.

BUCK EVAL iset ncon

iset = Load set

ncon = Constraint case

Created in processor EIG

NJ = 1

NI = Number of eigenvalues

Contains frequency eigenvalues corresponding to each eigenvector in "BUCK MODE."

VIBR MODE iset ncon

iset = Load set

ncon = Constraint case

Created in processor EIG.

NJ = Number of joints

NI = 6 minus number of joint motion components constrained on START card Number of blocks = Number of eigenvectors

Contents:

Each block of data contains one eigenvector corresponding to an eigenvalue stored in "VIBR EVAL." Data is stored for each joint in each active direction.

VIBR EVAL iset ncon

iset = Load set

ncon = Constraint case

Created in processor EIG.

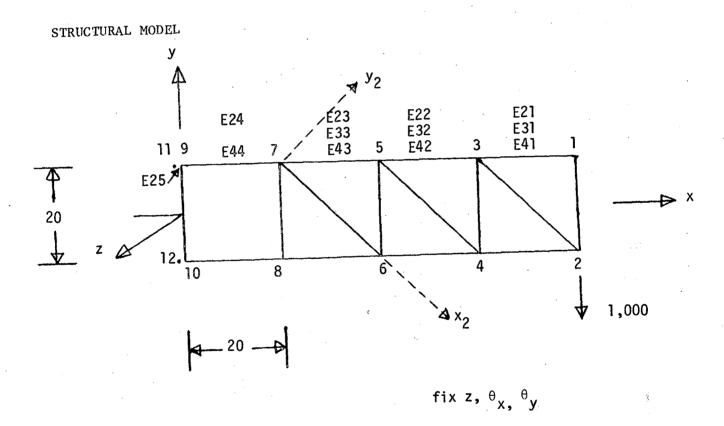
NJ = 1

NI = Number of eigenvalues

Contains eigenvalues corresponding to each eigenvector in "VIBR MODE."

EXAMPLE PROBLEM

The sample problem contained in this section includes the necessary SPAR commands to create all the data sets documented in this report. The SPAR commands are listed so that the user may reproduce these data sets and examine their contents. The resulting table of contents is included to illustrate the SPAR's use of data sets.



SPAR INPUT FILE

```
CXGT TAB S
START 12 3,4,5 S
TEXT 5
  EXAMPLE TO ILLUSTRATE DATA SET CONTENTS
  PREPARED BY GARY L. GILES 8-13-81.
MATC 5
               0.3 U.283 H.3=6 S
       30.0+6
NSW 5
    1 23.0 $
ALTREP S .
    2 1 0.0 2 0.0 5 -45,0 20.0 10.0 0.0 $
JLOC >
   1 80.0 10.0 0.0 0.0 10.0 0.0 5 2 $
    2 80.0 -10.0 0.0 0.0 -10.0 0.0 5 2 5
  11
       0.4 10.0 0.0 $
  12
     0.6 -10.0 0.0 $
JREF S
   NHEF#217 S
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   FORMAIR1 5
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    2 1 1000.0 0.0 0.0 $
BRL S
    1 1 0.0 1.0 0.0 1 0.0 1.0 0.0 $
BA S
           1.0 0.1 0.5 0.05 $
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BB 5
    1 11.0+3 8
      41.0
             $ 8+0.55
      31.0
             32.0
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SA S
    NMATEL 5
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       0.01 $
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CON#1 5
    2EHU 1,2,11,12 $
JSEQ &
    1/10,12,11 $
RMASS &
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        100 1.0,2.0,3.0 $
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LXGT TUPO S
IXQT K $
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LXQT AUS S
    ALPHAICASE TITL 1 S
IN SAMPLE TITLE FOR CASE 1 3
  SYSVELJAPPLIED FURCES 1 $
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CASE 1 3
    J=2;J=2;-1000.0 5
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    1=2; J=1; -2.0 5
 TABLEINGGAL TEMPERATURES 2 $
    J=1,4;100,0 5
 TABLEINGCAL PRESSURES 2 &
    Jal, 10,1,5 5
 ELDATATIEMP E41 2 1 5
   G=1; E=1; 100.0, 200.0, 300.0, 400.0 $
 ELDATAIDISL EST 2 1 5
    G=1;E=1,2;0.1,0.2,0.3 $
 ELDATAPPRES E43 2 1 8
    G=1,t=1,0.1.0.2,0.3,0.4 $
EUNF S
 RESET SET=2 5
IXOT SSUL &
[XGT GSF &
 RESET EMBEDES $
IXUT PSF 5
LXGT NG S
CXQT ELG S
 RESET PROBBBUCK, INIT#15, NOYNET, NREG#2 $
IXGT M 5
IXOT EIG &
 RESET PROBEVIOR, INITES, NOYNET, NEGES &
EXGT DCU #
   TCC 1
IXGT EXIT
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1	7	810820	151843	ø	18	1	18	Ü	JDF1	BTAB		
5	-8	810820	151843	ű	12	چ ړ	12	ŏ	JREF	BTAB	5	8
3	- 9	059018	151843	0	šš	1	12	1	ALTR	BTAB	5	4
4	10	058018	151843	0	30	Š	30	4	TEXT	BTAB	2	1
5	11	010820	151843	0	10	1	10	1	MATC	BTAB	5	á
6	12	058018	151843	()	1	1	1	ī	NSW	HTAB	Ş	3
7	13	058018	151843	U	5 4	å	24	1	ALTR	BTAB	5	4
8	14	010820	151843	0	36	12	36	ĩ	JLOC	BTAB	2	5
9	15	050018	151843	U	15	12	12	ő	JREF	BTAB	Ş	
10	16	6,10850	151843	0	10	2	10	1	MHEF	BTAB	Ž	6
11	17	810850	151843	Ú)	8	1	8	1	BRL	BTAB	ā	ė
12	18	610850	151843	0	31	1	31	1	HA	BTAB	2 2	9
13	19	056018	151643	0	21	1	21	1	88	BTAB	ž	10
14	50	810850	151843	0	6	1	6	1	H C	BTAB	2	11
15	4 1	010020	151843	Ú	6	1	6	1	80	BTAB	Ž	12
16	2.5	810820	151843	0	43	1	43	1	SA	BTAB	2	13
17	23	058018	151843	0	Ü	1	4	1	S 8	BTAB	2	14
18	8 4	020018	151643	()	12	12	12	Ø	CON		1	Q
19	25	010820	151843	Ů	1 2	5 !	12	0	JSEQ	BTAB	5	17
0.5	20	910850	151843	0	36	12	36	1	HMAS	BTAB	2	18
21	27	010820	151843	U	108	15	108	1	GJJT	BATB	2	19
25	29	058018	151958	0	54	49	882	0	DEF	E21	1	2
23	43	810820	151950	0	2	1	2	0	GD	E21	1	2 2 2 3 3 3 3 3 3 3
24	44	810850	151950	0	15	1	15	4	GTIT	E21	1	2
25	45	058018	151958	Û	20	1	20	0	DIR	E21	1	2
26	46	058018	151958	0	30	59	885	0	DEF	E31	6	3
27	60	050019	151958	0	2	1	S	O	GD -	E31	6	3
85	61	010820	151958	0	15	1	15	4	GTIT	E31	, 6	3
29 30	53	050019	151958	0	50	1	0 5	O	DIR	E31	6	
	63	059019	151958	0	16	56	896	0	DEF	E41	9	4
31	77	058018	151958	0	2	1	Š	0	GD	E41	9	4
32 33	78 65	058018	151958	0	15	, J	15	4	GTIT	E41	9	4
34	86	810820	151958 151958	0	, 50	l a	0.5	0	DIR	E41	9	4
35	100	810820	151958	0	54	49	588	0	DEF	ESS	5	2
36	101		151958	0	5	1	2	0	GD	E22	2	5
37	102	810820	151950	0	15 20	1	15	4	GTIT		2	2
38	103	810650	151958	0	30	5 O	. 20	0	DEF	E22	5	2
39	117	810820	151958	0	30	59	885 2	0	GO	£32	7	5
40	118	810820	151958	ő	15	1			GTIT	E32	7	3
41	119		151956	Ó	50	1	15 20	4	DIR	£32	7	2 2 3 3 3 3 4
42	120		151958	0	16	1 56	896	0	DEF	E42	7 10	3
43	134		151958	0	5	1	2	0	GD	E42		
44		810820	151958	Ü	15	1	15	4	GTIT	E42	10 10	4
45	136		151958	Ü	50	1	50	0	UIR	E42	10	4
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40	137	910950	151958	0	5 4	49	885	Q.	DEF	£23	3	2
47	151	810820	151958	Ò	ê	1	. 2	0	GD	E 23	3	Ş
48	156	810820	151958	Ü	15	•	15	4	GTIT			
49	153	810820	151950			A.				E23	3	2
				0	Šõ	- 1	05	0	DIR	E23	3	2
5 u	154	810850	151958	0	30	59	885	Ø	UEF	£33	e	3
5 1	168	810820	151958	Ü	2	1	5	U	GD	E33	8	3
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53	170	810820	151958	U	20	1	5.0	Û	DIR	£33	ě	1
54	171	610850	151958	ů,	10	56	696	•	DEF			3
55	185	610950	151958					Ú		E43	11	: 4
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56	106	010050	151958	0	15	į	15	4	GTIT	E43	11	4
57	167	010020	151958	0	50	1	50	Q	DIR	E43	11	4
58	100	020018	151958	U	7 2	49	888	0	DEF	E24	4	2
59	202	050018	151958	()	2	1	2	Ô	GO	E24	4	Ž
60	203	010850	151958	Ó	15	1	15	4	GTIT	E24	4	5
61	204	610850	151958	Ğ		4			-			5
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65	205	810820	151958	U	16	56	896	Q	DEF	E44	12	4
63	619	810850	151958	0	2	1	5	0	GD	E 4 4	12	4
64	550	610820	151958	0	15	1	15	4	GTIT	E 44	12	4
65	227	810820	151958	0	20	1	50	Ú	DIR	E44	1 2	4
66	828	058018	151958	Ü	36	49	882	Ö	DEF	E 25	5	2
67	242	810820	151958	0	ž		2	ő	GD			
68	243	910950	151958			a.		-		E25	5	2
				0	15	1	15	4	GTIT	£ 25	5	2
69	244	028018	151958	U	20	1	50	Q	DIR	E 25	5	2
70	245	810820	151958	0	1 2	12	12	4	ELTS	NAME	0	0
7 1	246	050018	151958	U	15	12	12	U	ELTS	NNOD	0	0
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73	248	028018	151958	Ö	180	12	180	0	NS	2001		
74	251	910950	152019								0	0
				0	384	3	128	4	E21	EFIL	1	2.
75	257	910850	152019	0	256	2	158	4	E31	EFIL	6	3
76	501	810850	152019	0	195	1	105	4	E 4 1	EFIL	9	. 4
77	604	610850	152019	0	304	3	120	4	E22	EFIL	2	2
78	075	020018	152019	0	384	2	105	4	E32	EFIL	7	3
79	276	010820	152019	0	256	1	256	u	E42	EFIL	10	ผ
80	280	020018	152019	ő	.384	3	128	4	E23	EFIL		
81	982	810820							-		3	Š
			152019	0	512	Ş	256	4	E33	EFIL	8	3
85	294	010820	152019	0	448	1	448	4	E43	EFIL	11	4
83	301	810820	152019	0	512	4	128	4	E 24	EFIL	4	2
84	369	010820	152019	Ü	128	1	158	4	E44	EFIL	12	U
85	311	810820	152019	Ü	256	2	128	4	£25	EFIL	5	2
86	315	810820	152009	0	36	SI	36	- 1	DEM	DIAG	ō	ō
87	316	810820	152024	Ö	896	12	896	Ö	KMAP	W W	35	10
68	330	028018										
			152024	O	1792	12	1792	0	AMAP		37	10
89	358	810820	152027	0	2240	12	2240	1	K	SPAR	9	0
90	393	810950	152028	Ü	3584	12	3584	1	INV	K	1	0
91	449	810820	152030	0	15	1	15	4	CASE	TITL	1	1
92	450	058018	152030	0	36	12	36	- 1	APPL	FORC	1	1
93	451	058018	152030	0	. 36	12	36		APPL	MOTI		i
94	452	810820		ő	12	12	12	-1	NODA	TEMP	5	•
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96	454	910960	152030	U	4	1	4	• 1	TEMP	E 4 1	2	1

97	461 610650	152030	()	- 6	2	6	- 1	UISL	E31	2	1
98	158018 504	152030	0		1	Ц	- 1	PRES	E43	Ş	1
99	463 810826	152033	0	16	3	18	e į	15	E 2 1	ä	1
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105	358018 604	152033	()	18	ž	18	- 1	IS	E22	2	1
103	467 81082	152033	9	1 2	Š	Ši	- <u>1</u>	18	£32	5	•
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108	472 610826		0	24	ű	24	- 1	ĪS	E34	5	•
109	473 610820		0	6	1	6	- 1	18	E44	2	•
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113	477 610820		0	36	15	36	- 1	STAT	REAC	4	ę t
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CONCLUDING REMARKS

The SPAR structural analysis system consists of a collection of processors for performing finite element analysis. The data generated by each of these processors is stored in a data file organized as a "library." Each data set, containing a two-dimensional table or matrix, is identified by a four-word name listed in the table of contents.

This report documents the contents of many of the SPAR data sets stored in the data base complex. The creating SPAR processor, number of rows and columns, and definitions of each of the data items, are listed for each data set. An example SPAR problem using these data sets is also presented. The SPAR input for the problem and resulting table of contents are included, so that the user may reproduce these data sets. This information can be used (1) to understand more clearly and to use more productively the existing processors in the SPAR system, (2) to develop new SPAR processors, or (3) to interface SPAR with another software system.

REFERENCES

- 1. Whetstone, W. D.: SPAR Structural Analysis System Reference Manual System Level II. Volume I Program Execution. NASA CR-145096-1, 1977.
- Giles, Gary L.; and Haftka, Raphael T.: SPAR Data Handling Utilities. NASA TM-78701, September 1978.

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Washington, DC 20546				•				
15. Supplementary Notes								
16. Abstract				,				
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