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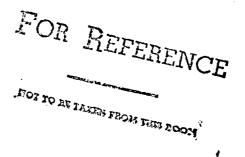
NASA Contractor Report 172452

ICASE REPORT NO. 84-46

ICASE

NASA-CR-172452 19850001995

REVIEW OF SOME VORTEX RELATIONS



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Contract No. NAS1-17070 September 1984

INSTITUTE FOR COMPUTER APPLICATIONS IN SCIENCE AND ENGINEERING NASA Langley Research Center, Hampton, Virginia 23665

Operated by the Universities Space Research Association



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Space Administration

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1 1 RN/NASA-CR-172452 3 DISPLAY 03/2/1 85N10302*# ISSUE 1 CATEGORY 34 PAGE 46 RPT#: NASA-CR-172452 ICASE-84-46 NAS 1.26:172452 CNT#: NAS1-17070 84/09/00 10 PAGES UNCLASSIFIED DOCUMENT UTTL: Review of some vortex relations TLSP: Final Report AUTH: A/KRAUSE, E. CORP: National Aeronautics and Space Administration. Langley Research Center, Hampton, Va. AVAIL.NTIS SAP: HC A02/MF A01 MAJS: /*COMPRESSIBLE FLOW/*DAMPING/*INCOMPRESSIBLE FLOW/*INVISCID FLOW/*MOMENTUM **/*VORTICES** MINS: / CIRCULATION/ COMPUTATION/ INTEGRALS/ KINETIC ENERGY/ VISCOUS FLOW ABA: Author ABS: The evaluation of the circulation from numerical solutions of the momentum and energy equations is discussed for incompressible and compressible flows. It is shown how artificial damping directly influences the time ratio of change of the circulation.

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REVIEW OF SOME VORTEX RELATIONS

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Abstract

The evaluation of the circulation from numerical solutions of the momentum and energy equations is discussed for incompressible and compressible flows. It is shown how artificial damping directly influences the time rate of change of the circulation.

Research was supported by the National Aeronautics and Space Administration under NASA Contract No. NASI-17070 while the author was in residence at ICASE, NASA Langley Research Center, Hampton, VA 23665.

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Introduction

Most finite-difference approximations of the Euler equations and of the Navier-Stokes equations for large Reynolds numbers must resort to artificial damping in order to stabilize the solution of the difference equations. Since the discretization process itself and the introduction of the damping terms changes the differential balance of mass, momentum, and energy, it is of general interest to know to what extent the local flow behavior and the overall flow properties are affected. If, for example, unsteady vortex formation is to be predicted, the net spacing and the time steps must be appropriately chosen, or else the numerical solution may completely suppress the formation of vortices. Or, if solution branching occurs, the results predicted may not correspond to the correct branch at all [1]. Another example is the prediction of inviscid transonic flows. Severe falsification of the total pressure may result in the computation if either the truncation error is too large or the damping terms are not adequately modelled. In fact, for irrotational flows the total pressure is often taken as a measure of accuracy of the numerical solution.

For rotational flows, in particular for unsteady flows, such a measure is not always available. It is then advantageous, to trace back the influence of the artificial damping on the integral properties of the solution. As the circulation is a characteristic integral property for rotational flows, its dependence on the local flow properties is reviewed here in some detail. First the material derivative of the circulation is discussed in conjunction with the Thomson theorem. Thereby the overall influence of artificial damping terms on the circulation can be demonstrated. By making use of various forms of the energy equation, it is then shown how for compressible flows the material derivative of the circulation is related to the entropy, heat conduction, and dissipation.

2. The Material Derivative of the Circulation

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As is well known (see, for example [2]), the material derivative of the circulation Γ is obtained by considering its change during an infinitesimal interval of time dt:

$$\frac{\mathrm{d}\Gamma}{\mathrm{d}t} = \oint_{\mathbf{C}} \frac{\mathrm{d}\overline{\mathbf{v}}}{\mathrm{d}t} \cdot \mathrm{d}\overline{\mathbf{s}} \quad . \tag{2.1}$$

Equation (2.1) can be written in a different form by splitting up the material derivative of the velocity into its local and convective part and by eliminating the latter with the vector identity

$$(\overline{\mathbf{v}} \cdot \nabla) \overline{\mathbf{v}} = \nabla \left(\frac{\mathbf{v}^2}{2} \right) - \overline{\mathbf{v}}_{\mathbf{x}} (\nabla \mathbf{x} \overline{\mathbf{v}}) \,. \tag{2.2}$$

There results

$$\frac{\mathrm{d}\Gamma}{\mathrm{d}t} = \oint_{C} \frac{\partial \overline{v}}{\partial t} \cdot \mathrm{d}\overline{s} - \oint_{C} \left(\overline{v} x (\nabla x \overline{v}) \right) \cdot \mathrm{d}\overline{s}.$$
(2.3)

This relation is valid for incompressible and compressible flows, as is equation (2.1). Another way of eliminating the material derivative of the velocity in equation (2.1) is to replace it through the momentum equation

$$\frac{d\overline{v}}{dt} = -\frac{1}{\rho} \nabla p - \frac{1}{\rho} \nabla \cdot \sigma^{-} . \qquad (2.4)$$

In equation (2.4) the quantity σ' represents the Stokes stress tensor without the pressure. Equation (2.1) then yields

$$\frac{\mathrm{d}\Gamma}{\mathrm{d}t} = -\oint_{c} \frac{\mathrm{d}p}{\rho} - \oint_{c} \frac{1}{\rho} (\nabla \cdot \sigma') \cdot \mathrm{d}\overline{s}. \qquad (2.5)$$

For incompressible flow the first line integral in equation (2.5), representing the change of the static pressure along the closed curve c, vanishes. Then $d\Gamma/dt$ is solely determined by viscous forces, i.e.

$$\frac{\mathrm{d}\Gamma}{\mathrm{d}t} = -\nu \oint_{\mathbf{c}} (\nabla^2 \ \overline{\mathbf{v}}) \cdot \mathrm{d}\overline{\mathbf{s}}.$$
(2.6)

Equation (2.6) contains Thomson's theorem: if the kinematic viscosity vanishes, the material derivative of the circulation is zero, i.e. $d\Gamma/dt = 0$, and the circulation remains constant, if it is initially constant. From equation (2.3) it follows for $d\Gamma/dt = 0$ that

$$\oint_{c} \frac{\partial \overline{v}}{\partial t} \cdot d\overline{s} = \oint_{c} (\overline{v}_{x}(\overline{v}_{x}\overline{v})) \cdot d\overline{s}.$$
(2.7)

Equation (2.6) can serve to evaluate the time development of the circulation in numerical solutions for incompressible flow.

3. Compressible Flows

The material derivative of the circulation can also be obtained for compressible flow. Since the density is then not constant along the curve c, the first integral in equation (2.5) does not vanish. The differential of the pressure in equation (2.4) can be expressed by the differentials of enthalpy and entropy according to the second law of thermodynamics,

$$\frac{1}{\rho} dp = dh - Tds, \qquad (3.1)$$

and equation (2.5) becomes

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$$\frac{\mathrm{d}\Gamma}{\mathrm{d}t} = \oint_{\mathrm{c}} \mathrm{T}\mathrm{d}\mathrm{s} - \oint_{\mathrm{c}} \left(\frac{1}{\rho} \nabla \cdot \sigma^{-}\right) \cdot \mathrm{d}\overline{\mathrm{s}}. \tag{3.2}$$

The line integral over dh vanishes for the same reasons given in conjunction with equation (2.6). For inviscid flows equation (3.2) reduces to [2]

$$\frac{\mathrm{d}\Gamma}{\mathrm{d}t} = \oint_{\mathrm{C}} \mathrm{T}\mathrm{d}s. \tag{3.3}$$

In contrast to incompressible flow, circulation can be generated in compressible inviscid flow, as long as the entropy and temperature vary along the path of integration in such a way that the line integral in equation (3.3) does not vanish.

Equation (2.3) can also be derived from equation (3.3) by using Crocco's theorem. It is obtained by expressing the static enthalpy h in equation (3.1) through the total enthalpy and the kinetic energy

$$dh = dh_0 - d \frac{v^2}{2}$$
, (3.4)

and by using the vector identity, equation (2.2), and the momentum equation, (equation (2.4))

$$T\nabla s = \nabla h_0 + \frac{\partial \overline{v}}{\partial t} - \overline{v} x (\nabla x \overline{v}) + \frac{1}{\rho} \nabla \cdot \sigma^{-}. \qquad (3.5)$$

This relation is identical with Crocco's theorem if the term describing the viscous forces is set equal to zero. If equation (3.5) is inserted in equation (3.2), it is seen that equation (2.8) is recovered.

In numerical computations the material derivative of the circulation can be evaluated either from equation (2.5), equation (3.2) or, for inviscid flows, from equation (3.3). The influence of heat conduction and energy dissipation can be demonstrated by differentiating equation (3.4) with respect to time, and by inserting the energy equation into the resulting relation. With

$$\frac{dh_0}{dt} = \frac{1}{\rho} \frac{\partial p}{\partial t} + \frac{1}{\rho} \nabla \cdot (\lambda \nabla T - \sigma \cdot \overline{v}), \qquad (3.6)$$

there is then obtained ·

$$T \frac{ds}{dt} = \frac{1}{\rho} \nabla \cdot (\lambda \nabla T) - \frac{1}{\rho} (\sigma' : \nabla \overline{\nu}) . \qquad (3.7)$$

A direct substitution of this relation into equation (3.2) is, however, not possible.

4. Applications

Some of the relations discussed can be used for numerical evaluation. If artificial damping is introduced, the momentum equation, (equation (2.4)), may contain two additional terms of second and fourth order:

$$\frac{d\overline{v}}{dt} = -\frac{1}{\rho} \nabla p - \frac{1}{\rho} \nabla \cdot \sigma^{-} + f_{1} \nabla^{2} \overline{v} + f_{2} \nabla^{4} \overline{v}, \qquad (4.1)$$

where f_1 and f_2 designate the damping coefficients. The material derivative of the circulation is then

$$\frac{d\mathbf{r}}{d\mathbf{t}} = -\oint_{\mathbf{c}} \frac{d\mathbf{p}}{\rho} - \oint_{\mathbf{c}} \frac{1}{\rho} (\nabla \cdot \sigma^{*}) \cdot d\mathbf{\overline{s}}$$

$$+ \oint_{\mathbf{c}} f_{1}(\nabla^{2} \ \overline{\mathbf{v}}) \cdot d\mathbf{\overline{s}} + \oint_{\mathbf{c}} f_{2}(\nabla^{4} \ \overline{\mathbf{v}}) \cdot d\mathbf{\overline{s}}.$$

$$(4.2)$$

The last two integrals can be evaluated separately so that their contribution to the circulation can be checked as a function of time. The damping of the energy equation can be estimated from equation (3.7), if corresponding damping terms are added.

5. Conclusions

The evaluation of the circulation in numerical solutions was discussed for incompressible and compressible flows. It was shown that the material derivative of the circulation is directly influenced through artificial damping terms in the momentum equation, and indirectly through damping terms in the energy equation.

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| 1. Report No. NASA CR-172452 ICASE Report No. 84-46 | 2. Government Acce | sion No. | 3. Rec | ipient's Catalog No. | |
|--|---|---|--|----------------------|--|
| 4. Title and Subtitle Review of Some Vortex R | | 5. Report Date September 1984 6. Performing Organization Code | | | |
| 7. Author(s) Egon Krause | | 84-46 | 8. Performing Organization Report No. 84-46 10. Work Unit No. | | |
| 9. Performing Organization Name and Add Institute for Computer and Engineering | | 11. Con | tract or Grant No. | | |
| Mail Stop 132C, NASA La Hampton, VA 23665 12. Sponsoring Agency Name and Address | ter | 13. Тур | NAS1-17070 13. Type of Report and Period Covered Contractor Report | | |
| National Aeronautics an Washington, D.C. 20546 | tion | 14. Spor | nsoring Agency Code 31-83-01 | | |
| 15. Supplementary Notes Langley Technical Monit Final Report | or: J. C. South, | Jr. | | | |
| The evaluation of the energy equations is discu how artificial damping circulation. | | sible and | l compressible | flows. It is shown | |
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| 17. Key Words (Suggested by Author(s)) circulation artificial damping | | 18. Distribution Statement 34 - Fluid Mechanics & Heat Transfer 64 - Numerical Analysis Unclassified - Unlimited | | | |
| 19. Security Classif. (of this report) Unclassified | 20. Security Classif. (of this Unclassified | page) | 21. No. of Pages 9 | 22. Price AO2 | |

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