HOST INSTRUMENTATION R&D PROGRAM

OVERVIEW

by

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The HOST Instrumentation R&D program is focused on two main classes of instrumentation. The first class is for characterizing the environment around the turbine engine components. These instruments include those for measuring gas flows, gas temperatures, and heat fluxes. The second class of instruments is for characterizing the effect of the environment on the turbine engine components. The second class includes strain measurements and an optical system for viewing various other structural responses such as cracking, buckling, spalling, carbon buildup, etc.

The HOST Instrumentation R&D program was formulated to concentrate on the critical measurements that could not be made with commercially available instruments or with instruments that were already under development via NASA- or DOD-funded efforts or in IR&D programs.

The HOST Instrumentation R&D Program Schedule showing the current active efforts is included in the accompanying figures. The program schedule shows all HOST-funded efforts plus selected non-HOST-funded efforts initiated during the year prior to the start of the HOST program that directly relate to the HOST goals. Each line represents a separate contract, grant, or LeRC in-house effort.

The heaviest resources are concentrated on the measurements of strain and gas flow since these measurements are extremely critical to the success of the HOST program and the HOST requirements differ from the current state of the art by a considerable margin. Followup and complementary efforts not shown in the schedule are being planned for the strain measurement area.

TURBINE ENGINE HOT SECTION TECHNOLOGY

INSTRUMENTATION RESEARCH AND DEVELOPMENT

CS-82-2681

HOST INSTRUMENTATION R&D PROGRAM

GENERAL GOALS:

- DEVELOP INSTRUMENTATION FOR CHARACTERIZING THE ENVIRONMENT AROUND TURBINE ENGINE COMPONENTS
- DEVELOP INSTRUMENTATION FOR CHARACTERIZING THE EFFECT OF THE ENVIRONMENT ON THE TURBINE ENGINE COMPONENTS

CS-82-2680

HOST INSTRUMENTATION R&D PROGRAM

(SHOWING ACTIVE EFFORTS AS OF 10/82)

MEASUREMENT	FISCAL YEAR				GOAL	
	1980 1981	1982 1983	1984	1985		
VIEWING SYSTEM					HIGH-TEMPERATURE SYSTEM FOR VIEWING COMBUSTOR INTERIORS DURING OPERATION	
GAS TEMPERATURE					DYNAMIC GAS TEMPERATURE MEASUREMENT SYSTEM WITH 1 kHz RESPONSE	
STRAIN			_		TURBINE BLADE / VANE STATIC STRAIN GAGE	
					BURNER LINER STATIC STRAIN MEASUREMENT SYSTEM	
HEAT FLUX					BURNER LINER TOTAL HEAT FLUX SENSOR	
					TURBINE BLADE/VANE TOTAL HEAT FLUX SENS	
GAS FLOW					LASER ANEMOMETER SYSTEM FOR HIGH- PRESSURE HIGH-TEMPERATURE FLOWS	
					OPTIMIZATION OF LASER ANEMOMETER SYSTEMS	
	HOST FU	NDED /////	ION-HOST	FUNDED	CS-82-2682	

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TURBINE ENGINE HOT SECTION TECHNOLOGY

INSTRUMENTATION SESSION AGENDA

OVERVIEW	N. WENGER, LeRC
COMBUSTOR VIEWING SYSTEM	W. MOREY, UTRC
DYNAMIC GAS TEMPERATURE PROBE	D. ELMORE, P&W GPD
TURBINE BLADE/VANE STATIC STRAIN GAGE	F. LEMKEY, UTRC
LASER SPECKLE TECHNIQUE FOR BURNER LINER STRAIN MEASUREMENTS	K. STETSON, UTRC
HEAT FLUX SENSORS FOR BURNER LINERS AND TURBINE BLADES AND VANES	G. ALWANG, P&W CE
HOT SECTION LASER ANEMOMETRY	W. NIEBERDING, LeRC R. EDWARDS, CWRU

CS-82-2679