SURFACE PROTECTION OVERVIEW

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Turbine engine hot section materials are subjected to aggressive chemical and thermomechanical environments. High temperature environmental attack of dollar intensive turbine components reduces turbine efficiency and can limit life. The bottom line, of course, is that high temperature oxidation and hot corrosion attack costs you money. The objective of materials durability research at Lewis is to understand the mechanisms of alloy and coating attack, and the effects of interaction with the environment on mechanical behavior. This base of understanding provides the foundation for developing life prediction methods and identifying strategies for controlling attack via advanced metallic and ceramic coatings. The Turbine Engine Hot Section Technology Project (HOST) augments the life prediction area of our program.

Our objective under HOST is to develop a first-cut integrated environmental attack life prediction methodology for hot section components. Under HOST we are concerned with oxidation and hot corrosion attack of metallic coatings as well as their degradation by interdiffusion with the substrate. The effects of the environment and coatings on creep/fatigue behavior are being addressed through a joint effort with the Fatigue sub-project. Finally, an initial effort will attempt to scope the problem of thermal barrier coating life prediction. Verificiation of models will be carried out through benchmark rig tests including a 4 atm. replaceable blade turbine and a 50 atm. pressurized burner rig.

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National Aeronaut squand Space Administration

SURFACE PROTECTION OVERVIEW

S.R. LEVINE

MATERIALS DIVISION MATERIALS DURABILITY BRANCH

HIGH TEMPERATURE ENVIRONMENTAL ATTACK



SCHEMATIC OF MODES HIGH-TEMPERATURE ATTACK



STRESS

HOT CORROSION



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SURFACE PROTECTION

OBJECTIVE: DEVELOP AN INTEGRATED ENVIRONMENTAL ATTACK LIFE PREDICTION METHODOLOGY FOR HOT SECTION COMPONENT LIFE

APPROACH:

- COMPILE FIELD FAILURE MODES DATA BASE
- MODEL EFFECTS OF ENVIRONMENTAL ATTACK AND COATINGS ON CRACK INITIATION
- CORROSION/ERDSON MODEL FOR AIRFOILS
- COATING OXIDATION/DIFFUSION MODEL
- COATING HOT CORROSION LIFE PREDICTION
- THERMAL BARRIER COATINGS
- MODEL VERIFICATION / INTEGRATION

COMMENTS:

- THIS WILL BE A FIRST CUT
- TIME, MANPOWER AND ARE INSUFFICIENT UNDER HOST TO DEVELOP A FULLY SATISFACTORY METHODOLOGY

SURFACE PROTECTION

PROGRAM ELEMENT	FY81	82	83	84	85	86	87	EXPECTED RESULTS			
ENVIRONMENT/MECHANICAL PROPERTY INTERACTIONS (TASKS OF C/F CONTRACTS)						[I	MODEL FOR EFFECTS OF ENVIR ATTACK& COATINGS ON CRACK			
AIRFOIL DEPOSITION MODEL			<u> </u>			[MODEL TO PREDICT THE LOCATION &			
								ATTACK OF TURBINE AIRFOILS			
COATING LIFE PREDICTION								CAPABILITY TO PREDICT COATING DEGRADATION ON BLADES, VANES,			
								COMBUSTORS			

SURFACE PROTECTION



GOAL: DEVELOP THEORY TO PREDICT CORRODANT DEPOSITION ON TURBINE AIRFOILS DURATION: 36 MONTHS

APPROACH:

- GRANT DAN ROSNER, YALE
- EXTEND CHEMICALLY FROZEN BOUNDARY LAYER THEORY TO AIRFOILS
 - LAMINAR & TURBULENT FLOW
 - PRESSURE & TEMPERATURE OVER AIRFOIL

MULTI-COMPONENT CORRODANTS

AIRFOIL DEPOSITION MODEL VERIFICATION

- GOAL: VERIFY DEPOSITION MODEL AND INTEGRATE WITH ALLOY CORROSION RATE MODEL
- DURATION: 48 MONTHS, IN-HOUSE
- APPROACH: USING 4 ATM REPLACEABLE BLADE TURBINE RIG
 - VERIFY AIRFOIL DEPOSITION MODEL
 - VERIFY ALLOY RATE MODEL (FROM R&T BASE)
 - INTEGRATE TO LOCATION/RATE MODEL

COATING OXIDATION/DIFFUSION LIFE PREDICTION

GOAL: TO DEVELOP AN IMPROVED METHODOLOGY FOR PREDICTING OXIDATION LIFE OF METALLIC COATINGS

DURATION: 36 MONTHS, IN-HOUSE

APPROACH: SELECT, PROCURE COATED SPECIMENS DETERMINE COATING LIFE VS TEMPERATURE FURNACE

BURNER RIG

MEASURE SPALLING PARAMETERS FROM ISOTHERMAL OXIDATION DETERMINE KINETICS OF DIFFUSIONAL DEGRADATION DEVELOP CYCLIC OXIDATION MODEL WITH MODIFICATION FOR COATING DIFFUSIONAL DEGRADATION TEST MODEL BY VARIATION OF CYCLE FREQUENCY

HOT CORROSION SURFACE CHEMISTRY

GOAL: DETERMINE EFFECT OF SURFACE CHEMISTRY ON HOT CORROSION LIFE

DURATION: 36 MONTHS

- APPROACH: ANALYZE REPRESENTATIVE FIELD COMPONENTS
 - DETERMINE COATING LIVES FOR VARIATIONS IN

ALLOY

- COATING
- COATING AGE
- AGING METHOD
- MODEL RESULTS/SUGGEST METHODOLOGY FOR LIFE PREDICTION
- VERIFY METHODOLOGY

GOAL: CHARACTERIZE THE EFFECT OF COMBINED OXIDATION/HOT CORROSION CYCLIC EXPOSURE ON LIVES OF METALLIC COATINGS

DURATION: 36 MONTHS (IN-HOUSE)

APPROACH: DETERMINE MACH 0.3 BURNER RIG LIVES FOR COMBINED OXIDATION/ HOT CORROSION CYCLIC EXPOSURE

- VARIATIONS IN DEPOSITION RATE
- VARIATIONS IN TEMPERATURE LEVELS
- VARIATIONS IN COATING AGE

FIT RESULTS TO EMPIRICAL MODEL

LIFE PREDICTION VERIFICATION

OBJECTIVE: VERIFY COATING LIFE PREDICTION METHODOLOGY WITH BURNER RIG BENCHMARK TESTS

DURATION: 30 MONTHS

APPROACH: PREDICT COATING LIVES FOR SIMULATED LONG HAUL/SHORT HAUL COASTAL& INLAND MISSIONS

SIMULATE MISSIONS&MEASURE COATING LIFE

DIAGNOSE PREDICTIVE METHODOLOGY FOR DEFICIENCIES

THERMAL BARRIER COATINGS





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- GOAL: TO DEVELOP AN IMPROVED DESIGN/LIFE PREDICTION METHODOLOGY FOR THERMAL BARRIER COATINGS
- DURATION: 36 MONTHS
- APPROACH: COMPILE COATING PROPERTY DATA PREDICT COATING LIFE IN RIG& ENGINE

VARIATIONS IN:

COATING PROPERTIES COATING THICKNESS THERMAL CYCLE

TEST PREDICTIVE CAPABILITY BY RIG (& ENGINE) TEST

RIG/ENGINE CORRELATION

- GOAL: VERIFY CORROSION, EROSION AND COATING LIFE MODELS AT NEAR ENGINE CONDITIONS
- DURATION: 36 MONTHS (IN-HOUSE)

APPROACH: VERIFY MODELS USING HIGH PRESSURE BURNER RIG DEPOSITION/CORROSION METALLIC COATING LIFE THERMAL BARRIER COATING LIFE

> DETERMINE MODEL DEFICIENCIES ATTRIBUTABLE TO HIGH PRESSURE HIGH HEAT FLUX