#### SPACE STATION DEVELOPMENT WORK

SOLAR DYNAMIC SYSTEMS

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**NOVEMBER 13, 1984** 

#### SOLAR DYNAMIC SYSTEMS

WHAT IS IT?

WHY ARE WE INTERESTED?

WHAT ARE THE BENEFITS?

WHAT IS THE SOLAR DYNAMIC DEVELOPMENT PROGRAM?

### SYSTEM CONCEPT

SUN	MIRROR	RECEIVER WITH STORAGE MATL.	BRAYTON, RANKINE OR STIRLING HEAT ENGINE W/ALT.	SPACE RADIATOR	
HEAT SOURCE	HEAT COLLECTOR	HEAT STORAGE	HEAT TO ELECTRICITY CONVERTER	WASTE HEAT REJECTION	

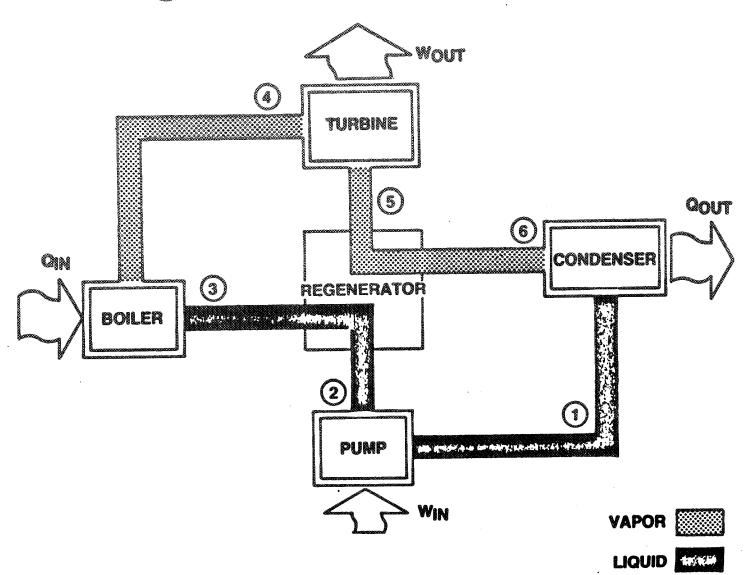
RADIATOR

BRAYTON, RANKINE OR STIRLING ENGINE W/ALTERNATOR

RECEIVER

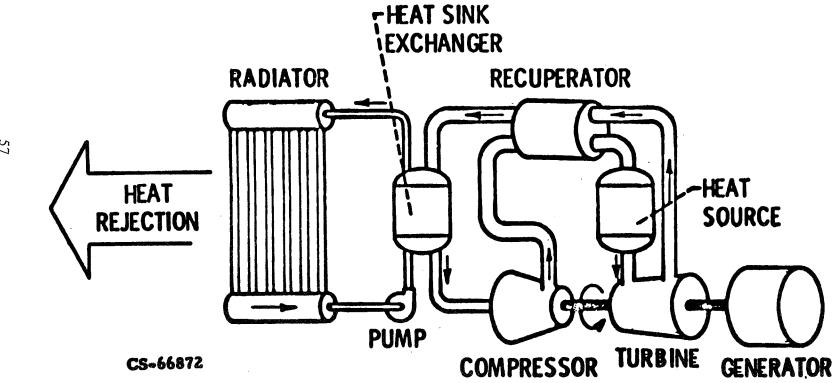
MIRROR

### Organic Rankine Cycle



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### BRAYTON CYCLE SPACE POWER SYSTEM



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### SOLAR DYNAMIC RADIATOR REQUIREMENTS

20 KW<sub>E</sub>

	ORGANIC RANKINE		BRAYTON			
TURBINE INLET TEMP	670K	(750F)	730K	(855F)	1080K	(1490F)
CONCENTRATOR DIA. AREA	14.6m 167m <sup>2</sup>		14.8m 170m <sup>2</sup>		12.6м 125м <sup>2</sup>	
HEAT REJECTION RATE	105 KW		115 KW		80 KW	
PRIME RADIATOR AREA	150m <sup>2</sup>		220м <sup>2</sup>		60m <sup>2</sup>	
RADIATOR TEMP	355K 344K	(180F) (160F)	376K 296K	(220F) (70F)	451K 324K	(350F) (120F)

### BENEFITS OF SOLAR DYNAMIC SYSTEMS OVER PV SYSTEMS

### **PRIMARY**

### SMALLER AREA

- DRAG

- STABILITY AND CONTROL
- ORBIT ALTITUDE ACCESSIBILITY
- VIEW ANGLE

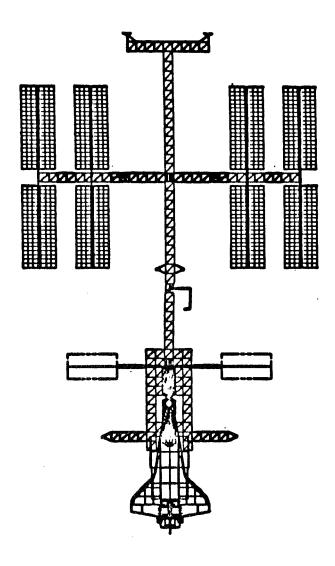
### SECONDARY

LOWER MASS

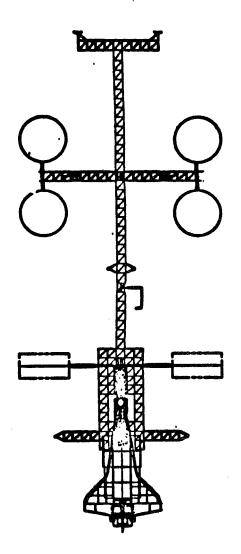
LOWER COST

HIGHER RELIABILITY

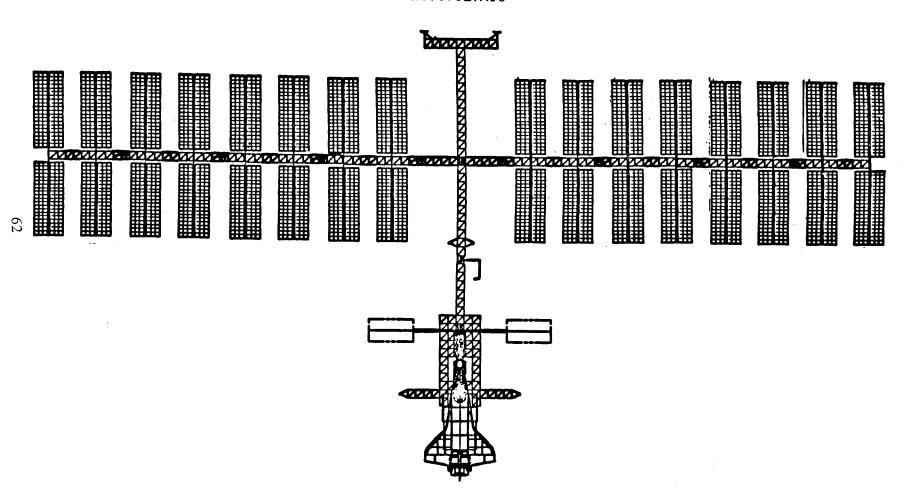
# 75 KW SPACE STATION PHOTOVOLTAIC



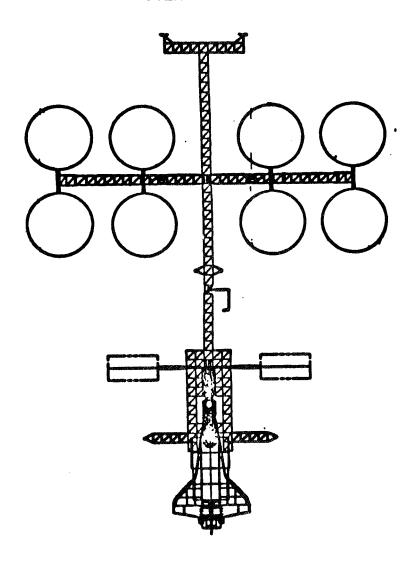
# 75 KW SPACE STATION SOLAR DYNAMIC



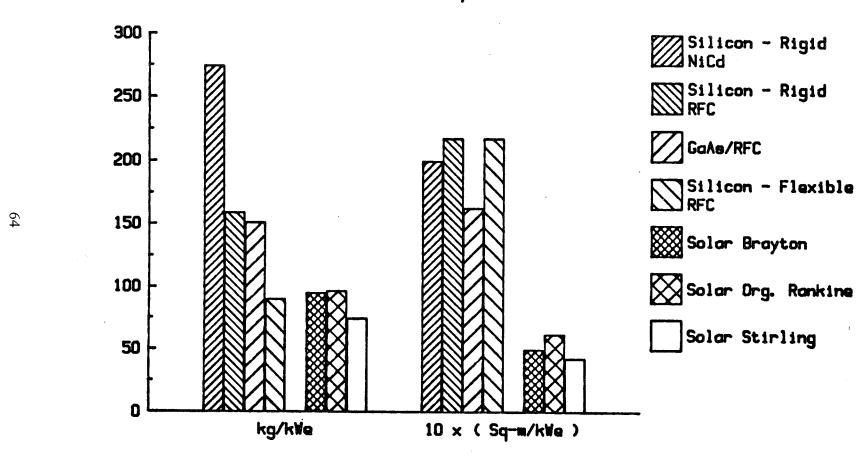
### 300 KW SPACE STATION PHOTOVOLTAIC



# 300 KW SPACE STATION SOLAR DYNAMIC



# SPECIFIC MASS AND AREA OF SEVERAL SPACE STATION POWER SYSTEMS Net Power Output = 75 kWe



LeRC Space Systems Office

### SOLAR DYNAMIC DEVELOPMENT HISTORY

- o SEVERAL SPACE CONCEPTS ACTIVELY PURSUED BY NASA AND USAF IN 1960's AND 70's
  - RANKINE
  - BRAYTON

1 - 15 KW<sub>E</sub> LEVELS

- KINEMATIC STIRLING
- O GREATEST EFFORT EXPENDED ON RANKINE AND BRAYTON POWER CONVERSION SUBSYSTEMS
- O LESS EFFORT EXPENDED ON CONCENTRATORS AND RECEIVERS
- O SYSTEMS DESIGNED FOR EXPENDABLE BOOSTER LAUNCH
- O NO SYSTEMS HAVE FLOWN
- O DEVELOPMENT DISCONTINUED IN EARLY 70'S FOR LACK OF SUITABLE MISSIONS
- O DOE DEVELOPING LOW COST CONCEPTS FOR TERRESTRIAL APPLICATIONS SINCE APPROX. 1977

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### SOLAR DYNAMIC ADVANCED DEVELOPMENT

#### **PURPOSE:**

PROVIDE A BASE OF DATA ON THE CRITICAL TECHNOLOGY OF SOLAR DYNAMIC SYSTEMS FOR SPACE STATION

- TO ASSESS VIABILITY OF THE SOLAR DYNAMIC OPTION FOR IOC
- TO PROVIDE BASIS FOR SYSTEM DESIGN

#### APPROACH:

- o ADDRESS PRIMARY CRITICAL TECHNOLOGY AREAS
  - RECEIVER/STORAGE
  - CONCENTRATOR
  - SOLAR DYNAMIC SYSTEM INTEGRATION WITH SPACE STATION
- o EMPLOY CONSERVATIVE DESIGN APPROACHES
- o FABRICATE AND TEST CRITICAL COMPONENT FUNCTIONS

### SOLAR DYNAMIC ADVANCED DEVELOPMENT

### CRITICAL CONCENTRATOR TECHNOLOGY

CENTERS: LERC, JPL

o DEFINE CONCENTRATOR REQUIREMENTS

o DEVELOP SEVERAL DESIGN CONCEPTS FOR 12-18 m. DIAMETER CONCENTRATORS

- FAB TECHNIQUES

- POINTING REQUIREMENTS

- ACCURACY

- SHUTTLE REQUIREMENTS

- DISTORTION

- STOWAGE

- SINGLE VS. CASSEGRAINIAN

- DEPLOYMENT/ERECTION

- COATINGS

o FABRICATE AND TEST ELEMENTS OF SELECTED DESIGNS

o TEST REFLECTIVE SURFACE SAMPLES IN SPACE

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### SOLAR DYNAMIC ADVANCED DEVELOPMENT

### CRITICAL HEAT RECEIVER TECHNOLOGY

CENTER: LERC

- DEFINE HEAT RECEIVER REQUIREMENTS
- DEVELOP SEVERAL DESIGN CONCEPTS FOR RECEIVERS IN THE RANGE OF 75-160 KW, AND TEMPERATURE NEAR 700K AND PERHAPS TO 1100K)
  - THERMAL STORAGE MATERIAL
- COATINGS
- CONTAINMENT OF STORAGE MATERIAL APERTURE MATERIAL
- MOVEMENT OF STORAGE MATERIAL
- HEAT PIPES
- TEMPERATURE CONTROL
- FABRICATION METHODS
- FABRICATE AND TEST DESIGNS OF ALTERNATE APPROACHES
- CAPSULE TESTS OF STORAGE AND CONTAINMENT MATERIALS