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Secondary Pattern Computation of an Offset Reflector Antenna

Roberto J. Acosta
Lewis Research Center
Cleveland, Ohio

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- ✓ Page 2, second paragraph, line 3 (line following eq. (1)): $\vec{f}(\theta, \phi)$ should be replaced by f .
- ✓ Page 2: Equations (1) and (4) should be enclosed in boxes.
- ✓ Page 3: Equations (5), (6), and (7) should be enclosed in boxes.
- Page 3, third paragraph, line 2: $0^r, \cdot$ should be $0^r, .$
- ✓ Page 4, second paragraph from bottom, line 4: Kuffman and Crowell should be Kauffman and Crowell.
- ✓ Page 4, second paragraph from bottom, line 13: Delete the following sentence: The agreement is also very good for the directivity.
- ✓ Page 11, reference 10: Clam, Peter T. should read Lam, Peter, T.C.
- Figure 6: The legend should read as follows: On-focus case (feed is at focal point).

SECONDARY PATTERN COMPUTATION OF AN OFFSET REFLECTOR ANTENNA

Roberto J. Acosta
National Aeronautics and Space Administration
Lewis Research Center
Cleveland, Ohio 44135

SUMMARY

Reflector antennas are widely used in communications satellite systems because they provide high gain at low cost. In analyzing reflector antennas the computation of the secondary pattern is the main concern. A computer program for calculating the secondary pattern of an offset reflector has been developed and implemented at the NASA Lewis Research Center. The theoretical foundation for this program is based on the use of geometrical optics to describe the fields from the feed to the reflector surface and to the aperture plane. The resulting aperture field distribution is then transformed to the far-field zone by the fast Fourier transform algorithm. Comparing this technique with other well-known techniques (the geometrical theory of diffraction, physical optics (Jacobi-Bessel), etc.) shows good agreement for large (diameter of 100λ or greater) reflector antennas.

INTRODUCTION

The accurate prediction of radiation characteristics for a microwave antenna is essential in designing antenna systems. Antenna radiation characteristics such as beam width, gain, aperture efficiency, side-lobe level, and cross polarization are used in analyzing and designing advanced antenna systems. The aperture field method (ref. 1) described in this report is one of several methods for predicting antenna performance characteristics. The method assumes that the tangential electric field on a planar surface in front of the antenna reflector is known. This aperture field distribution is then transformed to the far-field zone of the reflector by using a two-dimensional Fourier transform.

To compute the tangential electric fields on the aperture surface, the geometrical optics method (ref. 2) is used. In this technique the energy coming from the feed and reradiating from the reflector surface is characterized by an astigmatic tube of rays (ref. 3) that allows calculation of the amplitude and polarization of the electric field. This report also presents a method for computing the secondary pattern by using the aperture field method and a fast Fourier transform (ref. 4). Computation of the tangential electric field in the aperture plane is briefly described. A reflector configuration (fig. 1) is analyzed, and the results are compared with other well-known computational techniques. A description and a copy of the program are included in appendixes A and B, respectively.

The author wishes to thank Dr. Shung-Wu Lee and Dr. Peter T.C. Lam, of the University of Illinois, for their cooperation and recommendations.

APERTURE FIELD METHOD APPROACH

Description of Problem

The geometry of the problem under consideration is shown in figures 2 and 3. A reflector Σ_R is illuminated by the energy from a feed source at F_1 . The problem is defined as (1) to determine the tangential electric field at an observation point on the aperture grid F_2 , as shown in figure 2, and (2) to determine the secondary pattern from the near-field distribution by using the fast Fourier transform, as shown in figure 3.

Incident Electric Field on Reflector

The radiated electric field from the feed antenna has the asymptotic form given by equation (1):

$$\vec{E} \sim \frac{e^{-jkr}}{r} \vec{f}(\theta, \varphi) \quad (1)$$

where $\vec{f}(\theta, \varphi)$ is the active element pattern, $k = 2\pi/\lambda$ is the wave number, and r is the distance from the source to the reflection point. The vector function in equation (1) can be approximated by equation (2).

$$\vec{f}(\theta, \varphi) \cong \hat{\theta} U_E(\theta) (ae^{j\psi} \cos \varphi + b \sin \varphi) + \hat{\phi} U_H(\theta) (b \cos \varphi - ae^{j\psi} \sin \varphi) \quad (2)$$

where $U_E(\theta)$ is the E-plane active pattern, $U_H(\theta)$ is the H-plane active pattern, and a , b , and ψ are polarization parameters:

The various feed polarizations are described in the following table:

	a	b	ψ
Linear x	1	0	0
Linear y	0	1	0
Right-hand circular polarized (RHCP)	$1/\sqrt{2}$	$1/\sqrt{2}$	$\pi/2$
Left-hand circular polarized (LHCP)	$1/\sqrt{2}$	$1/\sqrt{2}$	$-\pi/2$

Typically these element patterns can be approximated by $(\cos \theta)^q$; that is,

$$U_E(\theta) = (\cos \theta)^{q_E} \quad (3a)$$

$$U_H(\theta) = (\cos \theta)^{q_H} \quad (3b)$$

If equations (3a) and (3b) are used to represent the active element patterns, the power radiated (ref. 5) by this source is given by equation (4).

$$P_{\text{RAD}} = \frac{q_E + q_H + 1}{60(2q_E + 1)(2q_H + 1)} \quad (4)$$

Two coordinate systems (fig. 4) are used, the feed coordinate system (x^F, y^F, z^F) and the reflector system (x, y, z) . These two coordinate systems are related by Eulerian angles (α, β, γ) illustrated in figure 5. Reference 6 provides a good description of these angles, which transform the incident field on the feed coordinate system into the reflector coordinate system.

Reflected Electric Field

For a given feed point F_1 and an observation point F_2 (fig. 2), a reflection point O^r may exist on the reflector Σ_R . This point is called the specular point. This type of ray reflection satisfies Snell's law of reflection. Reference 7 describes a search procedure to obtain the specular point. The reflected field at the aperture point F_2 is given by equation (5).

$$\vec{E}_{(F_2)}^r = DF \cdot e^{-jkd_2} [2(\hat{n} \cdot \vec{E}^i)\hat{n} - \vec{E}^i] \quad (5)$$

where d_2 is the distance from O^r to F_2 .

Equation (5) is given in terms of the incident field \vec{E}^i at the reflection point O^r , the surface unit normal \hat{n} of the reflector at O^r , and a divergence factor DF . The divergence factor in equation (5) is given by

$$DF = \frac{1}{\sqrt{1 + \frac{d_2}{R_1^r}}} \frac{1}{\sqrt{1 + \frac{d_2}{R_2^r}}} \quad (6)$$

where (R_1^r, R_2^r) are the principal radii of curvature of the reflected wavefront passing through O^r . Reference 7 gives an expression for these parameters as a function of the principal radii of curvature of a parabolic reflector.

Secondary Pattern

From the Hygens theorem (ref. 1) a solution for the far-field zone may be obtained if the tangential fields at Σ_A (fig. 3) are known. The aperture plane Σ_A is taken to be perpendicular to the z-axis.

The field in the far-field zone is given in terms of the tangential electric field at Σ_A , which is denoted by \vec{E}_a (ref. 8). Equation (7) relates the aperture field \vec{E}_a to the far-field zone electric field.

$$\vec{E}(\vec{r}) \sim jk_0 \frac{e^{-jkr}}{2\pi r} [\hat{\theta}E_\theta + \hat{\phi}E_\phi] \quad (7)$$

$$E_{\theta} = F_y \cos \varphi \cos \theta - F_x \sin \varphi \cos \theta \quad (7a)$$

$$E_{\varphi} = F_x \cos \varphi + F_y \sin \varphi \quad (7b)$$

$$\vec{F} = \iint_{\Sigma_A} \vec{E}_a e^{jk(ux+vy)} dx dy = F_x \hat{a}_x + F_y \hat{a}_y \quad (7c)$$

$$u = k \sin \theta \cos \varphi \quad (7d)$$

$$v = k \sin \theta \sin \varphi \quad (7e)$$

and θ and φ are the spherical coordinates of far-field point. The integral in equation (7c) is a double Fourier transform. The computer program developed uses a fast Fourier transform (FFT) algorithm to accomplish this task. References 9 and 10 describe in detail the use of the FFT algorithm to solve for the double Fourier transform.

The aperture field theory used to determine the secondary pattern is exact if the tangential fields are known everywhere on the aperture plane Σ_A . When employing the FFT, Σ_A is truncated. To minimize the amount of computer time, Σ_A should be as small as possible while capturing almost all of the field. The number of grid points should satisfy the Nyquist (sampling) theorem. The secondary pattern can be reconstructed from a discrete set of near-field values if the spacing is $\lambda/2$ or less. Reference 11 describes sample spacing for high-gain reflector antennas.

This method of calculating the secondary pattern is accurate in cases where the antenna diameter is of the order of 100λ or more. If the antenna diameter is less than 100λ , the accuracy is reduced, specifically in the side-lobe region. This approach was originally used by ~~Kuffman and Crowell~~ (ref. 4) and developed extensively by many others (e.g., Hwang, et al., ref. 12). The reflector configuration described in figure 1 was analyzed by using various methods (physical optics (Jacobi-Bessel) (refs. 13 and 14), the geometrical theory of diffraction (ref. 15), and geometrical optics). The calculated patterns and directivities are shown in figures 6 and 7. Figure 6 shows an on-focus case; the directivity and the far-field pattern are in very good agreement. Figure 7 shows an off-focus (or scan) case; the agreement is still very good. The computer program given in appendix B was used to analyze this configuration. ~~The agreement is also very good for the directivity.~~ Appendix A contains a detailed description (users guide) of all the input parameters, including those for the configuration shown in figure 1.

Directivity

The far-field zone is usually divided into two orthogonal polarizations. Following Ludwig's definition 3 (ref. 16), the following unitary polarization vectors are introduced:

$$\hat{R} = \hat{\theta}(ae^{j\psi} \cos \varphi + b \sin \varphi) + \hat{\phi}(-ae^{j\psi} \sin \varphi + b \cos \varphi) \quad (8a)$$

$$\hat{C} = \hat{\theta}(ae^{-j\psi} \sin \varphi - b \cos \varphi) + \hat{\phi}(-ae^{-j\psi} \cos \varphi + b \sin \varphi) \quad (8b)$$

If the secondary pattern can be expressed as

$$\vec{E} = \frac{e^{-jkr}}{r} [\hat{\theta}E_{\theta} + \hat{\phi}E_{\phi}] \quad (9)$$

the reference-polarization expression for \vec{E} is

$$\vec{E} \cdot (\hat{R}^*)^* \quad (10a)$$

and the cross-polarization expression is

$$\vec{E} \cdot (\hat{C}^*)^* \quad (10b)$$

The directivity for the reference polarization is defined by

$$D_R(\theta, \varphi) = \frac{4\pi |\vec{E} \cdot \hat{R}|^2 / Z_0}{P_{\text{rad}}} \quad (11a)$$

Similarly directivity for the cross polarization is defined by

$$D_C(\theta, \varphi) = \frac{4\pi |\vec{E} \cdot \hat{C}|^2 / Z_0}{P_{\text{rad}}} \quad (11b)$$

CONCLUDING REMARKS

A method, called the aperture field technique, has been developed for calculating the secondary pattern of an offset reflector illuminated by a feed with arbitrary polarization. By using the fast Fourier transform the far-zone electric field is computed very efficiently. The results for the secondary pattern are in good agreement with those obtained by other well-known techniques. This method can be conveniently extended to secondary pattern computation for multiple and numerically specified reflectors.

The computer program based on the aperture field technique is one of the main research tools used at the NASA Lewis Research Center for analyzing advanced antenna systems.

APPENDIX A

DESCRIPTION OF PROGRAM

A computer program was designed to calculate the antenna far-field performance characteristics. The method of analysis is geometrical optics. The output (unit FT07) is interfaced with a plotting program to display the far-field values. Any of the available plotting routines in the computer library may be used. The main features in this program (SOURCE - RAP\$11) are the calculation of directivity, feed efficiency, and plots of E or H far-field plane cuts.

Input (Unit FT05) Parameters to RAP\$11

F focal length, m
F1 fraction of wavelength that defines surface sample spacing
F2 fraction of wavelength that defines aperture sample spacing
DIA diameter of reflector (if circular projection), m
XF,YF,ZF coordinates of center (origin) of feed coordinate system on reflector coordinate system
DELTX offset of center of reflector relative to x-axis, m
DELTY offset of center of reflector relative to y-axis, m
FREQ frequency
QE E-plane cosine pattern exponent
QH H-plane cosine pattern exponent
POL 1, Y-POL; 2, X-POL; 3, RHCP; 4, LHCP
ZETA maximum far-field angle for which pattern will be calculated

The input is described in figure 8.

The input for the case described in figures 1, 6, and 7 (to be read in FT05) is as follows:

F = 2.43	DELTY = 0.434
F1 = 0.5	FREQ = 11.74E9
F2 = 0.5	QE = 3.2
DIA = 2.754	QH = 2.6
XF = YF = ZF = 0.0	POL = 3
DELTX = -1.377	ZETA = 4.0

APPENDIX B

COMPUTER PROGRAM

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0000100 C*****THIS PROGRAM IS REFLECTOR ANALYSIS THIS THE MOST GENERAL VERSION
0000200 C*****THE METHOD OF ANALYSIS IS APERTURE INTEGRATION.
0000300
0000350     DIMENSION PNUM(500,500)
0000400     DIMENSION X(500),Y(500),ERX(500,500),ERY(500,500)
0000410     DIMENSION PH(500,500),EXR(5000),EXI(5000),EYR(5000),EYI(5000)
0000420     DIMENSION FIELDX(1000),FIELDY(1000),FIELD(1000),ANG(1000)
0000425     DIMENSION RTNARR(2)
0000500 C*****INPUT PARAMETERS TO THIS PROGRAM*****
0000501 C*****F : FOCAL LENGTH M
0000502 C*****F1 : FRACTION OF WAVELENTGH (SURFACE SAMPLE SPACING)
0000503 C*****F2 : FRACTION OF WAVELENTGH (APERTURE SAMPLE SPACING)
0000504 C*****DIA : DIAMETER M
0000505 C*****XF,YF,ZF : ORIGIN OF FEED COORDINATE SYSTEM.
0000506 C*****DELTX,DELTY : OFFSET X-AXIS,Y-AXIS RESP.
0000507 C*****FREQ : FREQUENCY HZ
0000508 C*****QE,QH,Q : COSINE PATTERN EXPONENT
0000509 C*****POL : 1 Y-POL, 2 X-POL, 3 RHCP, 4 LHCP
0000510 C*****ZETA : MAX FAR-FIELD ANGLE.
0000550     INTEGER*2 N1
0000600     NAMELIST/INPUT/F,F1,F2,DIA,XF,YF,ZF,DELTX,DELTY,FREQ,Q,ZETA,QH,
           QE,POL
0000700     READ(5,INPUT)
0000701 C*****ALL CONSTANT PARAMETEPS IN THE PROGRAM*****
0000710     PINT=0.
0000801     WRITE(9,997)
0000802 997  FORMAT(5X,'THIS IS RFLECTOR ANALYSIS PROGRAM. BY R.J. ACOSTA')
0000900     PI=4.*ATAN(1.)
0001000     AWAVE=3E8/FREQ
0001100     SSP=F1*AWAVE
0001200     NSP=(DIA/SSP)+1
0001300     SAP=F2*AWAVE
0001400     NAP=(DIA*1.5/SAP)+1
0001500     RMAX=0.5*DIA
0001600     XC=DELTX+RMAX
0001700     YC=DELTY+RMAX
0001710     ZC=-((XC**2+YC**2)/(4.*F))+F
0001800     XSMAX=XC+RMAX
0001900     YSMAX=YC+RMAX
0002000     XAMAX=XC+RMAX*1.5
0002100     YAMAX=YC+RMAX*1.5
0002110     ALIM=.5*SAP
0002120     FACT=((AWAVE**2)*(120.*PI))
0002130     FACT=1/FACT
0002131 C*****POWER RADIATED BY THE FEED SOURCE*****
0002140     PRAD=60.*(2.*Q+1)
0002150     PRAD=1/PRAD
0002151 C*****THIS CALCULATES THE SUBSTENTED ANGLE BY REFLECTOR REL TO FEED*****
0002160     XU=XC
0002161     YU=YC+RMAX
0002162     ZU=-((XU**2+YU**2)/(4.*F))+F
0002165     XL=XC
0002166     YL=YC-RMAX
0002168     ZL=-((XL**2+YL**2)/(4.*F))+F
0002169     A117=((XU-XF)*(XL-XF)+(YU-YF)*(YL-YF)+(ZU-ZF)*(ZL-ZF)
0002170     A115=((XU-XF)**2+(YU-YF)**2+(ZU-ZF)**2
0002171     A115=SQRT(A115)
0002175     A116=((XL-XF)**2+(YL-YF)**2+(ZL-ZF)**2
0002176     A116=SQRT(A116)
0002180     SI=ARCOS(A117/(A115*A116))

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0002181 C*****CALCULATION OF POWER INTERCEPTED*****
0002185 EXP=2.*Q+1
0002186 P11=-COS(.5*SI)**EXP
0002187 P12=P11/EXP+1/EXP
0002188 PINT=P12/60
0002189 PEFFR=PINT/PRAD
0002190 TAPPER=20.*ALOG10(COS(SI*.5)**Q)
0002195 WRITE(9,998)PEFFR,TAPPER
0002196 998 FORMAT(////,5X,'EFFICIENCY OF FEED',F10.3,5X,'TAPPER OF
FEED =',E15.5)
0002200 DO 400 I=1,NAP
0002210 DO 410 J=1,NAP
0002220 PNUM(I,J)=0.
0002230 ERX(I,J)=0.
0002240 ERY(I,J)=0.
0002250 410 CONTINUE
0002260 400 CONTINUE
0002300 C*****NUMERICAL INTEGRATION OF FEED POWER*****
0002309 DELTAP=PI/NSP
0002310 DELTA=SI/NSP
0002311 NSP1=NSP*.5
0002315 DO 201 I=1,NSP1
0002320 ZETA1=DELTA*(I-1)
0002325 A145=2.*Q
0002326 FACT1=COS(ZETA1)**A145
0002327 FACT1=FACT1*SIN(ZETA1)*DELTA
0002328 PINTN=PINTN+FACT1
0002330 201 CONTINUE
0002350 PINTN=PINTN/60
0002360 PNUR=PINTN/PINT
0002370 WRITE(9,993)PNUR
0002371 993 FORMAT(////,5X,'THIS NUMERICAL TO EXACT RATIO =',F10.3)
0002400
0002500
0002600
0002700
0002800 C*****SET SURFACE COORDINATES*****
0002900
0003000 DO 10 J=1,NSP
0003100 DO 11 I=1,NSP
0003200 XS=XSMAX-SSP*(I-1)
0003300 YS=YSMAX-SSP*(J-1)
0003400 RS=SQRT((XS-XC)**2+(YS-YC)**2)
0003500 IF(RS.LE.RMAX)GO TO 12
0003600 GO TO 11
0003700 12 CONTINUE
0003800 ZS=-((XS**2+YS**2)/(4.*F)+F
0003900 C*****CALCULATE THE NORMAL TO SURFACE*****
0004000
0004100 AMAGN=SQRT(XS**2+YS**2+(4.*F**2))
0004200 ANX=-XS/AMAGN
0004300 ANY=-YS/AMAGN
0004400 ANZ=-2.*F/AMAGN
0004500 C*****CALCULATE INCIDENT RAY*****
0004600
0004700 AIMAGN=SQRT((XS-XF)**2+(YS-YF)**2+(ZS-ZF)**2)
0004800 SIX=(XS-XF)/AIMAGN
0004900 SIY=(YS-YF)/AIMAGN
0005000 SIZ=(ZS-ZF)/AIMAGN
0005100 C*****CALCULATE REFLECTED RAY*****
0005200
0005300 SRX=SIX-2.*(SIX*ANX+SIY*ANY+SIZ*ANZ)*ANX
0005400 SRY=SIY-2.*(SIX*ANX+SIY*ANY+SIZ*ANZ)*ANY
0005500 SRZ=SIZ-2.*(SIX*ANX+SIY*ANY+SIZ*ANZ)*ANZ
0005600

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0005700 C*****APERTURE COORDINATE OF REFLECTED RAY*****
0005800
0005900      XA=X5-Z5*(SRX/SRZ)
0006000      YA=Y5-Z5*(SRY/SRZ)
0006100 C*****LOCATE THIS RAY IN THE APERTURE GRID MATRIX*****
0006200
0006300
0006400
0006500
0006700      DO 21 N=1,NAP
0006800      X(N)=XAMAX-SAP*(N-1)
0007000      XDELTA=ABS(X(N)-XA)
0007100      IF(XDELTA.LE.ALIM) GO TO 30
0007200 21      CONTINUE
0007210      GO TO 500
0007300 30      CONTINUE
0007400      DO 20 M=1,NAP
0007500      Y(M)=YAMAX-SAP*(M-1)
0007600      YDELTA=ABS(Y(M)-YA)
0007700      IF(YDELTA.LE.ALIM) GO TO 40
0007800 20      CONTINUE
0007810      GO TO 500
0007900 40      CONTINUE
0008000 C*****STAR THE CALCULATION OF FIELDS AT APERTURE*****
0008100
0008200      A11=SQRT((XC-XF)**2+(YC-YF)**2+(ZC-ZF)**2)
0008300      A12=SQRT((XS-XF)**2+(YS-YF)**2+(ZS-ZF)**2)
0008500      ADOT=(XC-XF)*(XS-XF)+(YC-YF)*(YS-YF)+(ZC-ZF)*(ZS-ZF)
0008600      ADOT=ADOT/(A11*A12)
0008612
0008700      FPAT=ADOT**Q
0008710 C****THIS IS TO CALCULATE THE TOTAL POWER INTERCEPTED BY REFLECTOR****
0008712
0008800      IF(FPAT.LT.1E-5)FPAT=1E-5
0008810 C*****FINISH FEED PATTERN CALCULATIONS*****
0008820 C*****INCIDENT UNIT VECTOR CALCULATION*****
0008900      EIX=-(SIX*SIY)
0009000      EIY=(SIX**2+SIZ**2)
0009100      EIZ=-(SIZ*SIY)
0009200      A13=SQRT(EIX*EIX+EIY*EIY+EIZ*EIZ)
0009300      EIX=EIX/A13
0009400      EIY=EIY/A13
0009500      EIZ=EIZ/A13
0009600 C*****INCIDENT FIELD AMPLITUDE*****
0009700      E1NX=EIX*FPAT/A12
0009800      E1NY=EIY*FPAT/A12
0009900      E1NZ=EIZ*FPAT/A12
0010000 C*****REFLECTED FIELD AMPLITUDE CALCULATION*****
0010100      ERFX=2.*(ANX*E1NX+ANY*E1NY+ANZ*E1NZ)*ANX-E1NX
0010200      ERFY=2.*(ANX*E1NX+ANY*E1NY+ANZ*E1NZ)*ANY-E1NY
0010300      ERFZ=2.*(ANX*E1NX+ANY*E1NY+ANZ*E1NZ)*ANZ-E1NZ
0010400 C*****THE PHASE CALCULATION *****
0010409      A31=SQRT((XA-XS)*(XA-XS)+(YA-YS)*(YA-YS)+ZS*ZS)
0010410      PHASE=(2*PI/AWAVE)*(A12+A31)
0010500      ERX(N,M)=ERX(N,M)+ERFX
0010600      ERY(N,M)=ERY(N,M)+ERFY
0010700      PH(N,M)=PH(N,M)+PHASE
0010800      PNUM(N,M)=PNUM(N,M)+1
0010900      GO TO 11
0011000 500      WRITE(9,300)
0011100 300      FORMAT(5X,'CAN NOT LOCATE THIS RAY????????????????')
0011200 11      CONTINUE
0011300 10      CONTINUE
0011301 C*****THIS WILL TAKE THE AVERAGE AMPLITUDE/PHASE*****
0011310      DO 60 I=1,NAP
0011315      DO 61 J=1,NAP

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0011318      IF(PNUM(I,J).EQ.0.) GO TO 61
0011320      ERX(I,J)=ERX(I,J)/PNUM(I,J)
0011321      ERY(I,J)=ERY(I,J)/PNUM(I,J)
0011322      PH(I,J)=PH(I,J)/PNUM(I,J)
0011325      61      CONTINUE
0011326      60      CONTINUE
0011400      C*****COLLAPSE THE FIELD BY ROWS OR IN X,*****
0011410      WRITE(55)NAP,NAP
0011417      WRITE(55)((ERY(I,J),J=1,NAP),I=1,NAP)
0011425      WRITE(55)((PH(I,J),J=1,NAP),I=1,NAP)
0011431      WRITE(55)((ERX(I,J),J=1,NAP),I=1,NAP)
0011437      WRITE(55)((PH(I,J),J=1,NAP),I=1,NAP)
0011500      C*****TO GET A COLUMN VECTOR OR Y DISTRIBUTION*****
0011600
0011700      DO 15 I=1,NAP
0011800      DO 16 J=1,NAP
0011820
0011825
0011900      EXR(I)=EXR(I)+ERX(J,I)*COS(PH(J,I))
0012000      EXI(I)=EXI(I)+ERX(J,I)*SIN(PH(J,I))
0012100      EYR(I)=EYR(I)+ERY(J,I)*COS(PH(J,I))
0012200      EYI(I)=EYI(I)+ERY(J,I)*SIN(PH(J,I))
0012300      16      CONTINUE
0012400      15      CONTINUE
0012401      C*****THIS IS TOTAL POWER IN X AND Y COPONETS OF N.F.*****
0012500      C*****APPLYING THE FFT OR APERTURE INTEGRATION METHOD*****
0012600      NAP=NAP*5
0012700
0012800      CALL FFT(EXR,EXI,NAP,NAP,NAP,1)
0012900      CALL FFT(EYR,EYI,NAP,NAP,NAP,1)
0013000      A133=NAP*SAP/AWAVE
0013100      NMAX=A133*SIN(ZETA*PI/180.)
0013200      NMAX=NMAX+1
0013300      DO 70 I=1,NMAX
0013400      J1=NMAX-(I-1)
0013500      A134=J1/A133
0013600      ANG(I)=-ARSIN(A134)*180./PI
0013700      J2=I+NMAX
0013800      A135=(I-1)/A133
0013900      ANG(J2)=ARSIN(A135)*180/PI
0014000      70      CONTINUE
0014100      DO 71 J=1,NMAX
0014200      I1=NAP-NMAX-1+J
0014300      FIELDX(J)=FACT*(EXR(I1)*EXR(I1)+EXI(I1)*EXI(I1))*COS(ANG(J)
          *PI/180)**2
0014400      FIELDY(J)=FACT*(EYR(I1)*EYR(I1)+EYI(I1)*EYI(I1))
0014500      FIELD(J)=FIELDX(J)+FIELDY(J)
0014600      I2=J+NMAX
0014700      FIELDX(I2)=FACT*(EXR(J)*EXR(J)+EXI(J)*EXI(J))*COS(ANG(J)
          *PI/180)**2
0014800      FIELDY(I2)=FACT*(EYR(J)*EYR(J)+EYI(J)*EYI(J))
0014900      FIELD(I2)=FIELDX(I2)+FIELDY(I2)
0015000      71      CONTINUE
0015100      N1=2*NMAX
0015200      CALL SCLBAK(.FALSE.,N1,FIELD,RTNARR)
0015300      GAIN=10.*ALOG10(4.*PI*RTNARR(2)*SAP**4/PRAD)
0015400      WRITE(9,950)GAIN
0015500      950     FORMAT(/////,5X,'THIS IS THE GAIN VALUE =',E15.5)
0015510      WRITE(9,965)N1
0015511      965     FORMAT(/////,5X,'TOTAL NUMBER OF CALCULATED FAR-FIELD
          VALUES =',I5)
0015600      DO 73 I=1,N1
0015700      FIELD(I)=FIELD(I)/RTNARR(2)
0015800      IF(FIELD(I).LT.1E-6)FIELD(I)=1E-6
0015900      FIELD(I)=10.*ALOG10(FIELD(I))
0016000      WRITE(7,565)ANG(I),FIELD(I)
0016100      565     FORMAT(5X,E15.5,5X,E15.5)
0016200      73      CONTINUE
0016300      STOP
0016400      END

```

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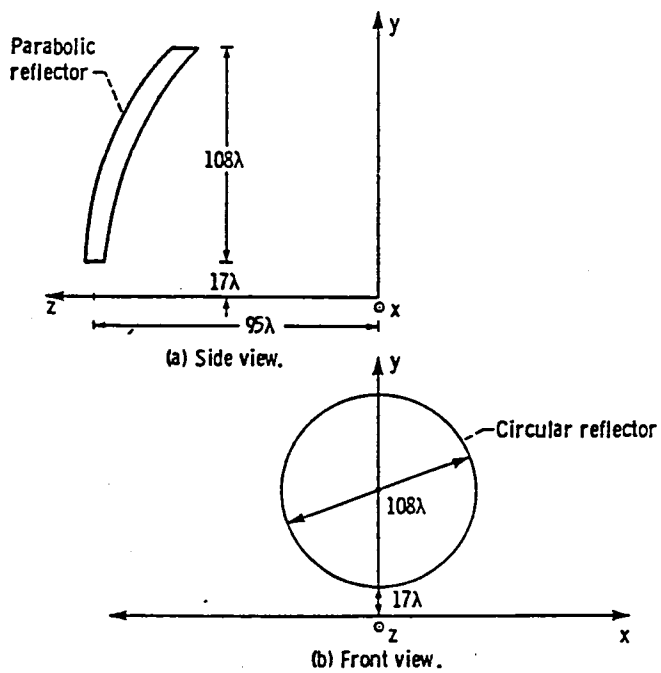


Figure 1 - Reflector configuration. Frequency, 11.75 GHz. Polarization of feed: $a = 1/\sqrt{2}$, $b = 1/\sqrt{2}$, and $\psi = 90^\circ$ for right-hand circular polarization; $U_E(\theta) = (\cos \theta)^{2.6}$; $U_H(\theta) = (\cos \theta)^{2.8}$.

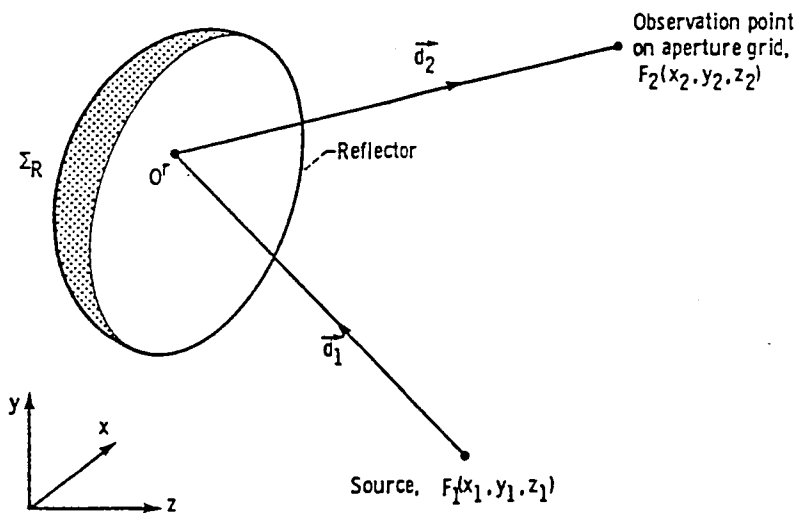


Figure 2. - Reflector Σ_R being illuminated by incident field from point source at F_1 .

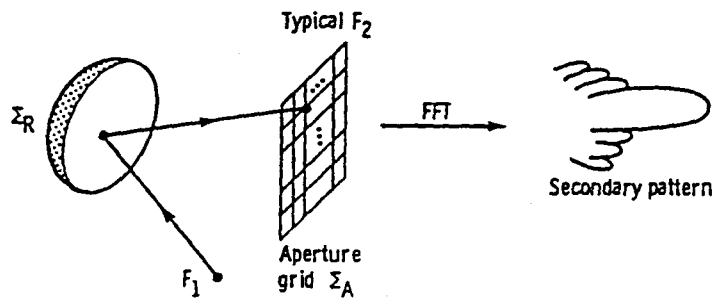


Figure 3 - Secondary pattern using fast Fourier transform (FFT).

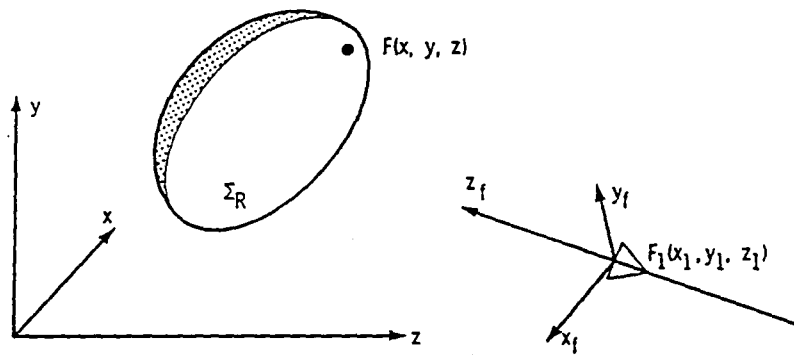


Figure 4 - Feed and reflector coordinate systems.

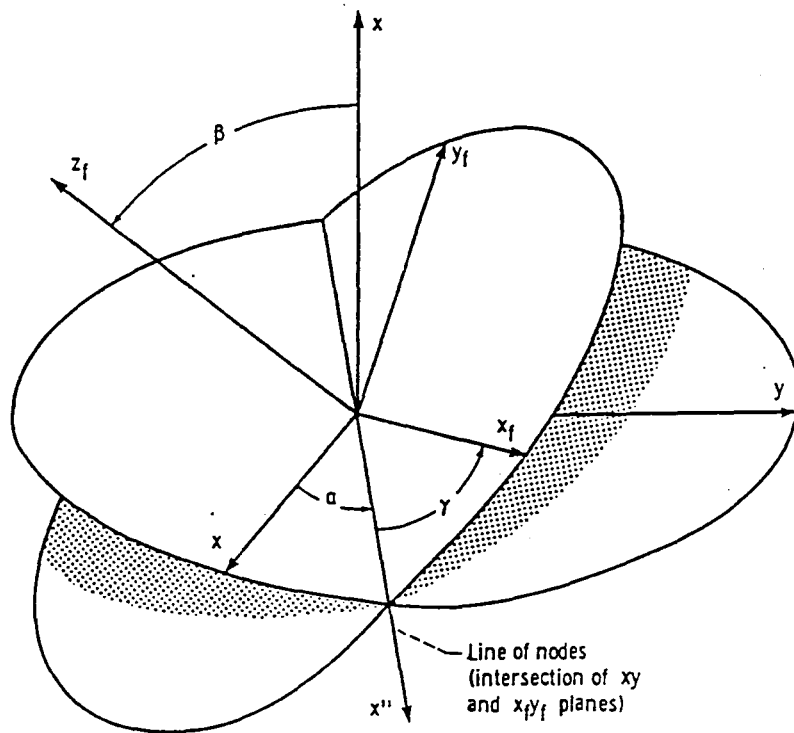


Figure 5. - Eulerian angles.

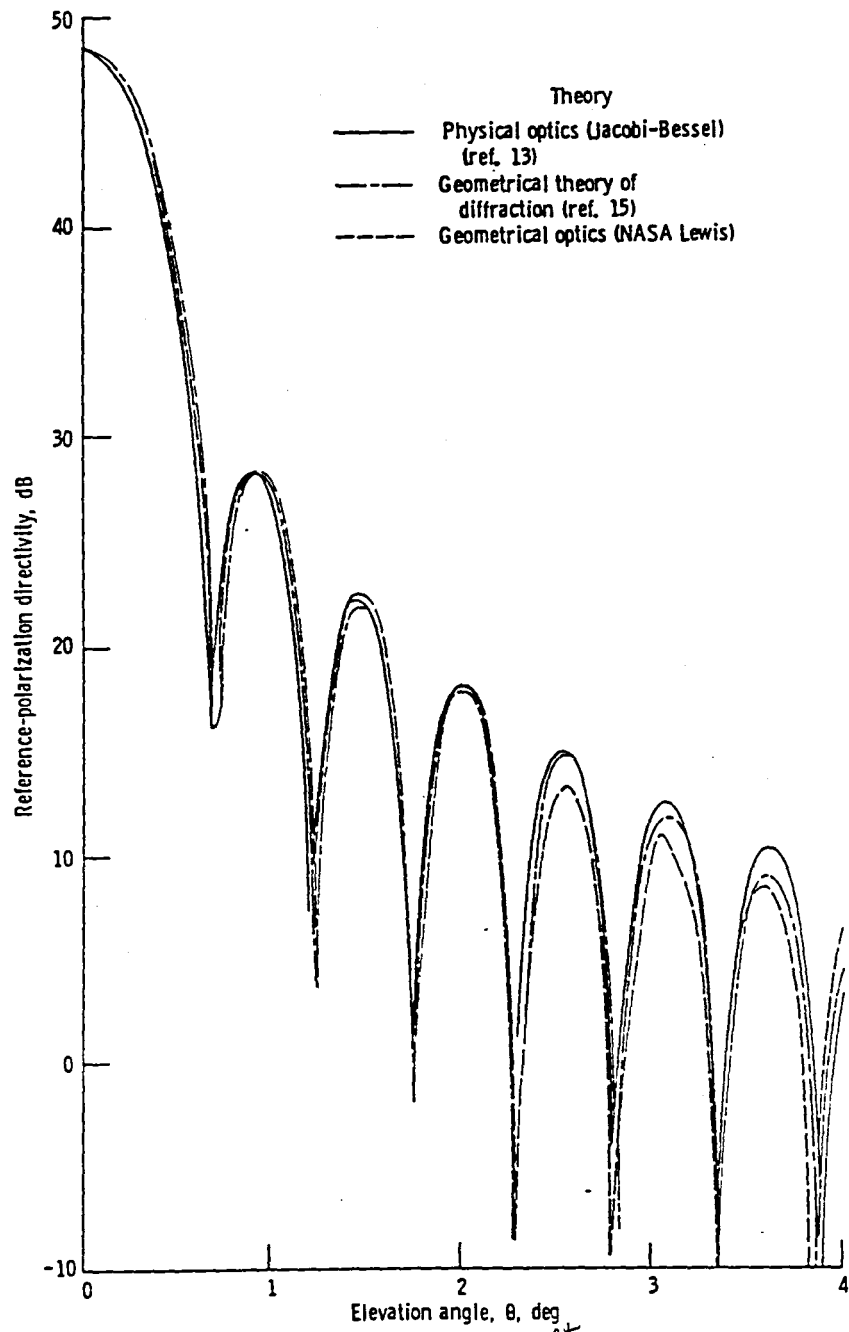


Figure 6. - On-focus case (feed is ^{at} focal point).

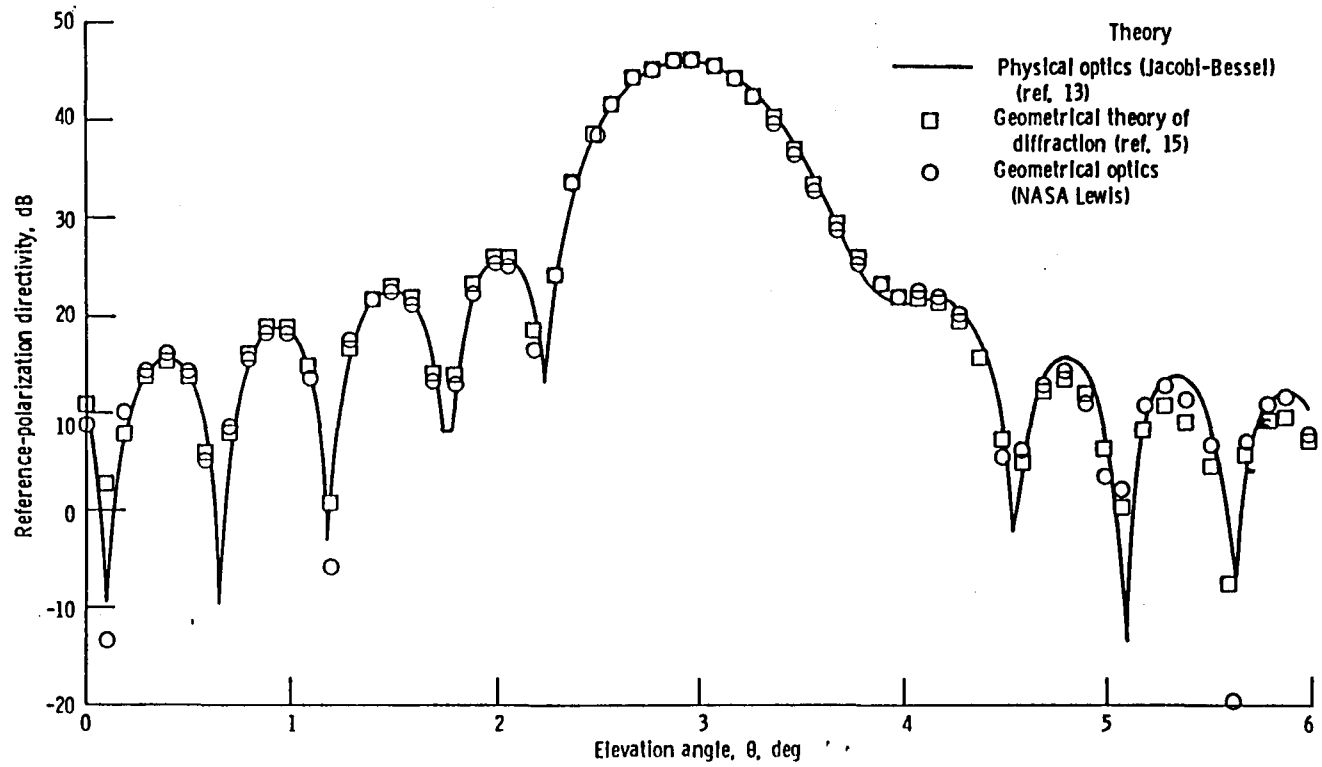
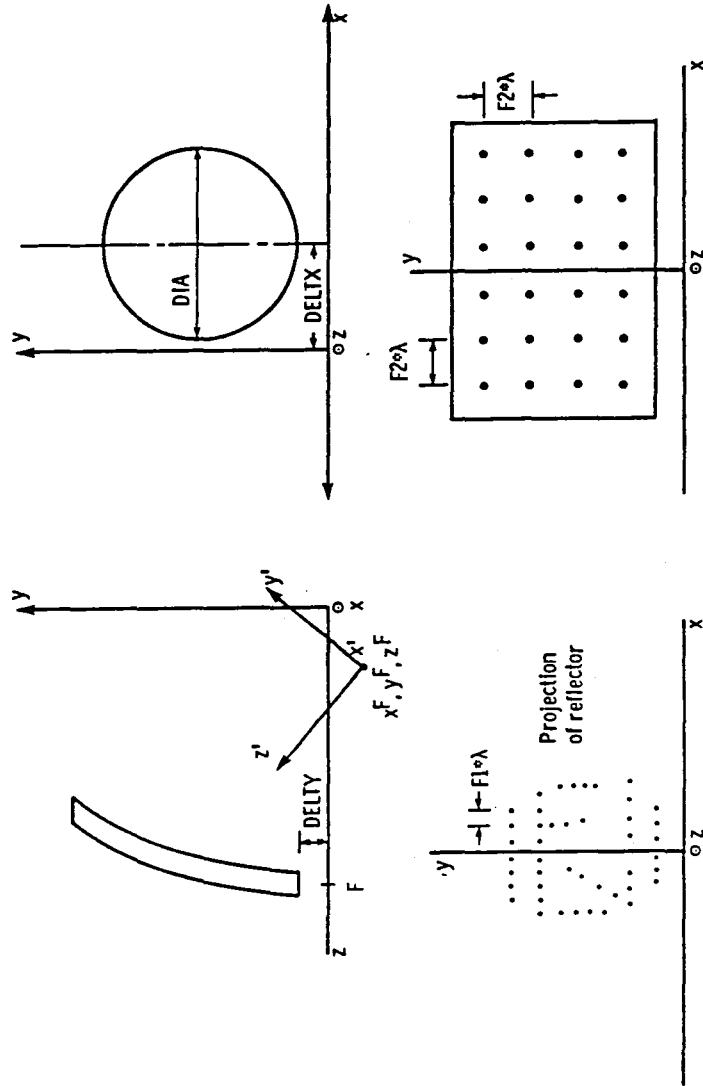


Figure 7. - Scan case (feed moved off focus).



(a) Surface grid.

(b) Aperture grid.

Figure 8. - Description of Input.

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16. Abstract <p>Reflector antennas are widely used in communications satellite systems because they provide high gain at low cost. In analyzing reflector antennas the computation of the secondary pattern is the main concern. A computer program for calculating the secondary pattern of an offset reflector has been developed and implemented at the NASA Lewis Research Center. The theoretical foundation for this program is based on the use of geometrical optics to describe the fields from the feed to the reflector surface and to the aperture plane. The resulting aperture field distribution is then transformed to the far-field zone by the fast Fourier transform algorithm. Comparing this technique with other well-known techniques (the geometrical theory of diffraction, physical optics (Jacobi-Bessel), etc.) shows good agreement for large (diameter of 100λ or greater) reflector antennas.</p>					
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