

IN-23699

2339.

DOE/NASA/0256-1
NASA CR-175083
MTI 86TR17

Performance of Oil Pumping Rings

An Analytical and Experimental Study

(NASA-CR-175083)	PERFORMANCE OF OIL PUMPING RINGS: AN ANALYTICAL AND EXPERIMENTAL STUDY	N86-31000
Final Report (Mechanical Technology, Inc.)		
233 p	CSSL 20D	Unclas
		G3/34 43508

M.W. Euseip, J.A. Walowit,
O. Pinkus, and P. Holmes
Mechanical Technology Incorporated

January 1986

Prepared for
NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
Lewis Research Center
Under Contract DEN 3-256

for
U.S. DEPARTMENT OF ENERGY
Conservation and Renewable Energy
Office of Vehicle and Engine R&D

DISCLAIMER

This report was prepared as an account of work sponsored by an agency of the United States Government. Neither the United States Government nor any agency thereof, nor any of their employees, makes any warranty, express or implied, or assumes any legal liability or responsibility for the accuracy, completeness, or usefulness of any information, apparatus, product, or process disclosed, or represents that its use would not infringe privately owned rights. Reference herein to any specific commercial product, process, or service by trade name, trademark, manufacturer, or otherwise, does not necessarily constitute or imply its endorsement, recommendation, or favoring by the United States Government or any agency thereof. The views and opinions of authors expressed herein do not necessarily state or reflect those of the United States Government or any agency thereof.

Printed in the United States of America

Available from

National Technical Information Service
U.S. Department of Commerce
5285 Port Royal Road
Springfield, VA 22161

NTIS price codes¹

Printed copy: A11

Microfiche copy: A01

¹Codes are used for pricing all publications. The code is determined by the number of pages in the publication. Information pertaining to the pricing codes can be found in the current issues of the following publications, which are generally available in most libraries: *Energy Research Abstracts (ERA)*; *Government Reports Announcements and Index (GRA and I)*; *Scientific and Technical Abstract Reports (STAR)*; and publication, NTIS-PR-360 available from NTIS at the above address.

DOE/NASA/0256-1
NASA CR-175083
MTI 86TR17

Performance of Oil Pumping Rings

An Analytical and Experimental Study

M.W. Eusepi, J.A. Walowit,
O. Pinkus, and P. Holmes
Mechanical Technology Incorporated
Latham, New York 12110

January 1986

Prepared for
National Aeronautics and Space Administration
Lewis Research Center
Cleveland, Ohio 44135
Under Contract DEN 3-256

for
U.S. DEPARTMENT OF ENERGY
Conservation and Renewable Energy
Office of Vehicle and Engine R&D
Washington, D.C. 20545
Under Interagency Agreement DE-AI01-85CE50112

TABLE OF CONTENTS

<u>SECTION</u>		<u>PAGE</u>
	NOMENCLATURE	v
	SUMMARY	1
1.0	INTRODUCTION	2
2.0	ANALYSIS WITH CONSTANT PARAMETERS	3
	2.1 Basic Equations	3
	2.2 Simplified Approach	7
	2.3 The Backstroke	9
	2.3.1 Analytical Approach	9
	2.3.2 Nature of Solution	11
	2.3.3 Performance Characteristics	15
3.0	ANALYSIS WITH VARIABLE PARAMETERS	25
	3.1 Thermal Effects	25
	3.1.1 The Energy Equation	25
	3.1.2 Lubricant Flow	26
	3.1.3 Modeling of Thermal Problem	33
	3.1.4 Calculation Procedure	35
	3.2 Variable Velocity and Squeeze-Film Effects	38
	3.3 Starvation	52
	3.4 Nonparallel Contours	54
4.0	PARAMETRIC STUDY	58

PRECEDING PAGE BLANK NOT FILMED

TABLE OF CONTENTS (CONT'D)

<u>SECTION</u>		<u>PAGE</u>
5.0	EXPERIMENTAL PROGRAM	74
	5.1 Tests with Large Babbitt Ring	77
	5.2 Tests with the Rulon and Carbon Ring	81
	5.3 Tests with the Small Babbitt Ring.	85
6.0	DISCUSSION OF RESULTS	86
	6.1 Empirical Correction for Starvation	87
	6.2 Effects of Viscosity	98
	6.3 Suggested Design Procedure	98
7.0	CONCLUSIONS	100
8.0	REFERENCES	101
	APPENDIX A: PRELIMINARY ANALYSIS OF PUMPING LENINGRADER SEAL.	102
	APPENDIX B: COMPUTER PROGRAM "RING"	116
	APPENDIX C: RESULTS OF EXPERIMENTS - NEW SERIES	181
	APPENDIX D: FIGURES SHOWING COMPARISON WITH EXPERIMENTAL DATA AND UNCORRECTED THEORY	203

NOMENCLATURE

A	Thermal constant, $\frac{6\mu_o U_o L}{\rho c_p C^2}$, (degree)
C	Radial clearance
C_M	Clearance at upstream edge when $\delta \neq 0$
d	Diameter
D	Flexural rigidity, $Et^3/[12(1 - \nu^2)]$
E	Elastic modulus
F	Radial shear force/unit circumferential length
\tilde{F}	Dimensionless shear force $FC^2/(6\mu_o U_o L^2)$
Hz	Hertz
K	Dimensionless flow rate, $\frac{Q}{\pi R U_o C}$
K_{eff}	Effective value of K corrected for starvation
L	Hydrodynamic land width
L_1	Effective distance from fixed end to start of film
\tilde{L}_1	Dimensionless length, L_1/L
$P(x)$	Radial loading function; see Equation (2-2)
Q	Volumetric flow rate
Q_o	Flow for $p_f = 0$
R	Radius of ring
T	Temperature
\tilde{T}	Dimensionless Temperature, $\frac{\rho c_p C^2 (T - T_o)}{6\mu_o U_o L}$
U_o	Average rod velocity
V	Normal velocity
c_p	Specific heat
e	Length over which p_o acts

NOMENCLATURE (CONT'D)

f	Frequency
h	Film thickness
h_1	Film thickness at $x = 0$
h_2	Film thickness at $x = L$
h_2'	Film thickness slope dh_2/dx at $x = L$
\tilde{h}	(h/C)
k	Integration constant
p	Hydrodynamic pressure
p_f	Sealed Pressure
p_{fm}	Maximum sealed pressure ($Q = 0$)
p_o	Clamping pressure
\tilde{p}	Dimensionless pressure $pC^2/(6\mu U_o L)$
\tilde{p}_f	Dimensionless sealed pressure $p_f C^2/(6\mu U_o L)$
s	Stroke
t	Ring thickness; time
u	Velocity
w	Elastic deflection
x	Position variable
x_c	Position variable at cavitation point
y	Variable across film
α	Geometric bending parameter, $\frac{t^2 (R + \frac{t}{2})^2}{12L^4 (1 - \nu^2)}$
β	Elastohydrodynamic parameter, $\frac{6\mu U_o L}{C^2} \frac{(R + \frac{t}{2})^2}{CtE}$
δ	Slope of tapered surface, $(C - C_M)/L$
ϵ	Dimensionless loading length, e/L

NOMENCLATURE (CONT'D)

η	Dimensionless film height, (y/h)
μ	Viscosity
μ_0	Inlet reference viscosity for thermal analysis
μ	Viscous heating function, rate of heat generated/unit area
ϕ	μ/μ_0
ρ	Density
ν	Poisson's Ratio
ξ	Dimensionless position variable x/L
σ	Squeeze film parameter (4L/S)
ψ	Stream function
λ	Dimensionless starvation factor

SUBSCRIPTS

c	Cavitation
F	Forward flow
E	Elastic
f	At x = L
m	Maximum
R	Back flow; backstroke
1	At leading edge
2	At trailing edge

SUMMARY

A steady-state design computer program has been developed to predict the performance of pumping rings as functions of geometry, applied loading, speed, ring modulus, and fluid viscosity. Additional analyses have been developed to predict transient behavior of the ring and the effects of temperature rises occurring in the hydrodynamic film between the ring and shaft. The analysis was initially compared with previous experimental data and then used to design additional rings for further testing.

Tests were performed with Rulon, carbon-graphite, and babbitt rings. Two different shaft diameters were used for the babbitt rings. The design analysis was used to size all of the rings and to select the ranges of clearances, thickness, and loading. Although full quantitative agreement was lacking, relative agreement existed in that rings that were predicted to perform well theoretically, generally performed well experimentally. Some causes for discrepancies between theory and experiment are believed to be due to starvation, leakage past the secondary seal at high pressures, and uncertainties in the small clearances and local inlet temperatures to the pumping ring.

The design criteria that evolved require the applied loading to be of the order of the desired pumping pressure, the flow requirements and tolerances to dictate clearance, and the elastic modulus and ring compliance to be such that the deflection under load statically results in clamping at very small interferences so that back flow is inhibited, but that excessive power loss and wear do not occur.

It was found that the pumping ring could be used to generate its own loading pressure without any priming if an initial taper was applied to the ring. However, for untapered rings, an initial loading had to be applied before self-pumping could be obtained.

A separate preliminary analysis has been performed for a pumping Leningrader seal. This analysis can be used to predict the film thickness and flow rate through the seal as a function of pressure, speed, loading, and geometry.

1.0 INTRODUCTION

An analysis of pumping rings was performed under certain simplifying conditions in previous phases of this work [1]. These conditions consisted of first ignoring the contribution, if any, of the back flow occurring during the return stroke of the rod. Other simplifications related to the use of constant or average parameters, namely constant viscosity and mean rod velocity, throughout the stroke. The latter also presumed the neglect of squeeze film forces due to the variation of film thickness engendered by the harmonic motion of the rod.

The comparison of experimental data with theoretical results based on the simplified analysis showed good agreement with respect to maximum pressures generated by the pumping ring. The flows produced at reservoir pressures below the maximum, however, were consistently lower in the experiments than those indicated by the analysis. The agreement between theory and experiment for the carbon-graphite rings [1] has been found to be incorrect due to an erroneous use of an excessively low viscosity in the theoretical computation.

The present work, an extension of the previous effort, is aimed at advancing the analysis of pumping rings. Thermal effects and variable rod velocity were included and, with it, the effects of squeeze film action in the fluid film. The analysis was extended to include the backstroke and concurrent cavitation and their effect on net flow. The new analysis was then used to run a parametric study in order to obtain optimized configurations of pumping rings of different shapes and materials including the effects of a geometric taper. The results of tests run on these optimized designs were then compared with calculations based on the new analysis.

A separate, preliminary analysis of the pumping Leningrader seal has also been performed. Since this analysis is separate from the pumping ring work, its results are presented in Appendix A.

2.0 ANALYSIS WITH CONSTANT PARAMETERS

2.1 Basic Equations

The equation governing deflection, w , of an axisymmetric shell under bending is

$$\frac{E t^3}{12(1-\nu^2)} \frac{d^4 w}{dx^4} + \frac{E t}{R^2} w = -P(x) \quad (2-1)$$

Referring to Figure 2-1, the radially outward loading, $P(x)$, may be expressed in terms of the clamping load, p_o , and the hydrodynamic pressure, p , as follows:

$$P(x) = \begin{cases} 0, & -L_1 \leq x \leq 0 \\ p, & 0 \leq x \leq L - e \\ p - p_o, & L - e \leq x \leq L \end{cases} \quad (2-2)$$

The hydrodynamic pressure, p , is determined from the solution of the Reynolds Equation

$$\frac{dp}{dx} = 6\mu U_o \left(\frac{h-k}{h^3} \right) \quad (2-3)$$

with the geometry of the hydrodynamic film as given in Figure 2-2. k is a constant of integration related to the flow, and U_o is the average speed, which is related to the frequency, f , and the stroke, s , by:

$$U_o = 2fs$$

This velocity represents the average of the sinusoidal velocity over each half cycle. If the system contains two opposing pumping rings, it is double acting and the average velocity, U_o is assumed to prevail over the entire cycle (though in two different pumping rings) for the forward stroke, as well as for the back-stroke. In this manner, the transient problem is reduced to and simplified into a quasi-steady-state process. For a single ring, the resulting flow should be divided by 2.

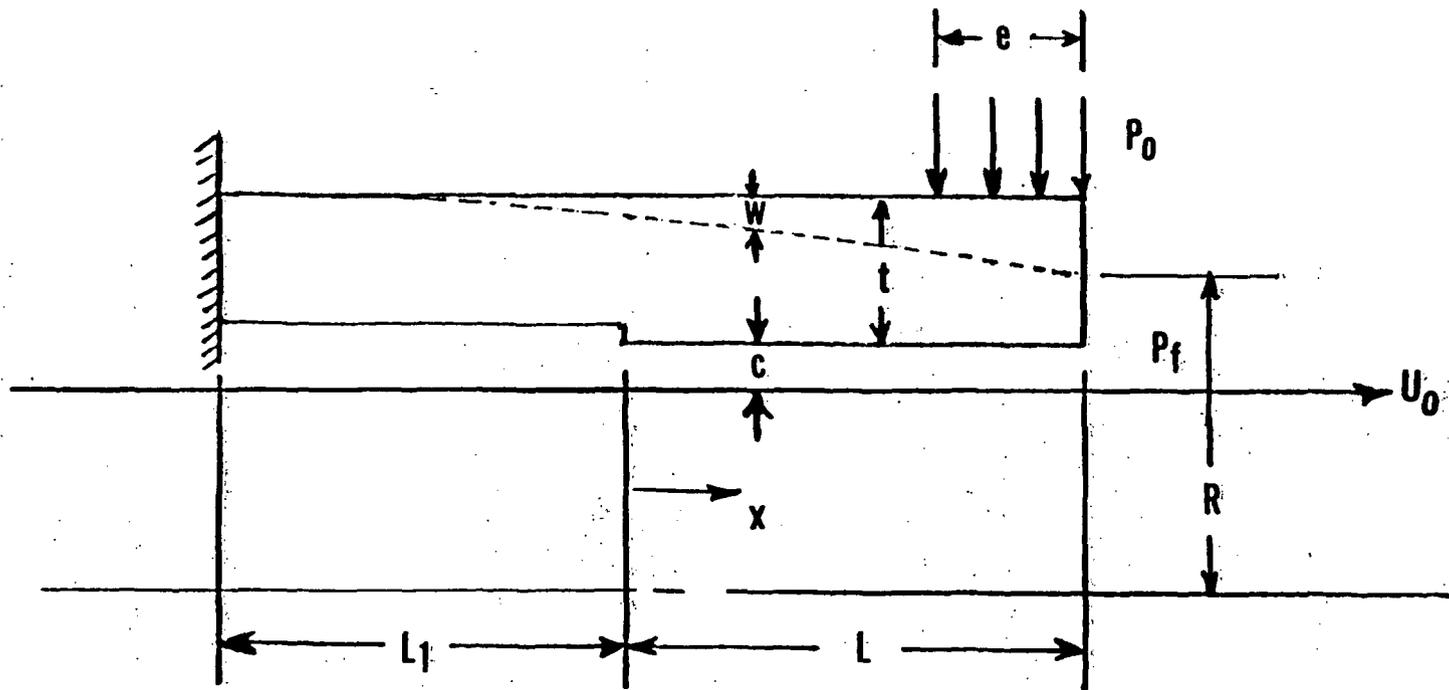


Fig. 2-1 Geometry of Pumping Ring

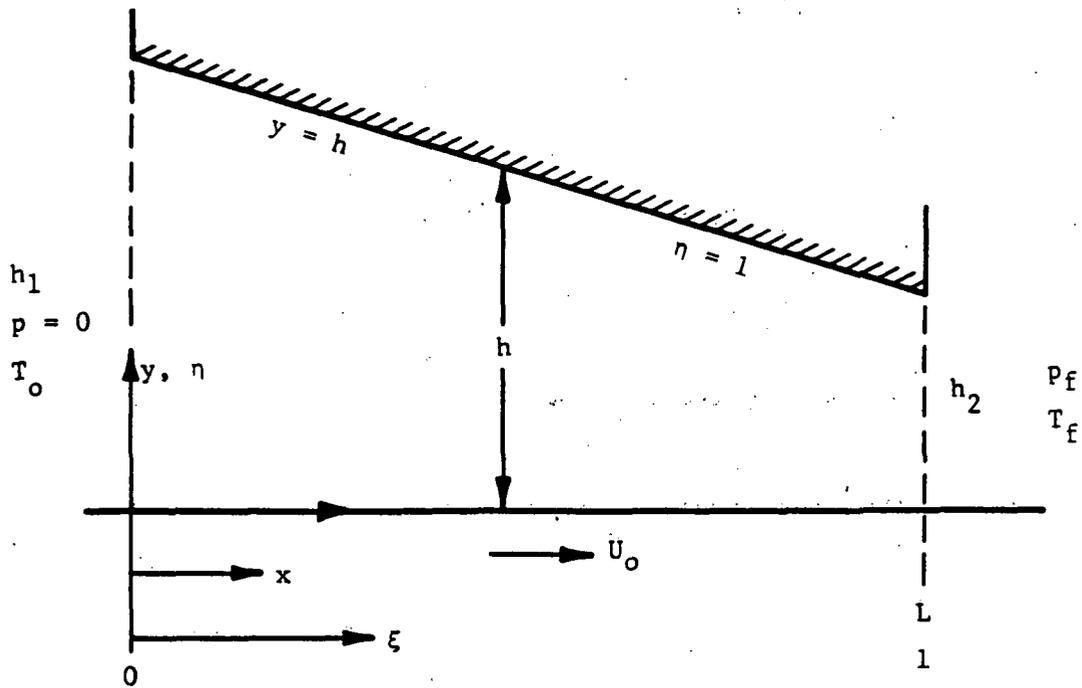


Fig. 2-2 The Hydrodynamic Film

The film thickness, h , appearing in Equation (2-3) is related to the deflection, w , by

$$h = (C - w), \quad 0 \leq x \leq L$$

The system of equations given by Equations (2-1) through (2-3) represents a fifth-order set of differential equations requiring five boundary conditions, in addition to a sixth condition for the evaluation of the constant, k , appearing in Equation (2-3). Two conditions result from the clamped-end requirement

$$w = \frac{dw}{dx} = 0 \text{ at } x = -L_1$$

Two conditions resulting from the prescribed pressures at $x = 0$ and L

$$p = 0 \text{ at } x = 0 \quad p = p_f \text{ at } x = L$$

where p_f is the sealed pressure. The remaining two conditions relate to the free-end requiring zero moment and zero shear, or

$$\frac{d^2w}{dx^2} = \frac{d^3w}{dx^3} = 0 \text{ at } x = L$$

The method of solving this set of equations subject to the specified boundary conditions is outlined in Reference [1]. The solution and the results of this analysis, reported in Reference [1], are based on the constancy of both viscosity and speed, namely

$$\begin{aligned} \mu &= \mu_0 = \text{constant} \\ U &= U_0 = 2fs = \text{constant} \end{aligned}$$

The analysis in Reference [1] did not consider the backstroke and was thus only applicable to sufficiently high loads and low elastic moduli to result in clamping during the backstroke.

2.2 Simplified Approach

A scrutiny of the pumping ring solutions formulated in Section 2.1 shows that the shape of the film over the hydrodynamic portion of the ring is nearly linear for the entire range of ring parameters and operating conditions. A few such examples are shown in Figure 2-3. Consequently, the problem can be considerably simplified if one postulates that the configuration of the film is tapered similar to that of a plane slider. In addition, a constant taper (shown in Figure 2-2) makes it possible to later treat the backstroke and accompanying cavitation. Mathematically a constant taper implies that $h' =$ constant. The pertinent expressions for the hydrodynamic component of pumping ring action are then given by

$$\tilde{h}(\xi) = \tilde{h}_2 + \Delta\tilde{h}(1 - \xi) \quad (2-4)$$

$$\tilde{p}(\tilde{h}) = 1/(\tilde{h}_2 - \tilde{h}_1) - 1/\tilde{h} + k'/(2\tilde{h}^2) + C_1 \quad (2-5)$$

where

$$k' = [2\tilde{h}_1\tilde{h}_2/(\tilde{h}_1 + \tilde{h}_2)][1 - (\tilde{h}_1\tilde{h}_2\tilde{p}_f)]$$

$$C_1 = [1/(\tilde{h}_1\tilde{h}_2)][1 + (\tilde{h}_2^2\tilde{p}_f)]$$

The elastic deformation equation remains the same and may be written in dimensionless form as

$$\alpha (d^4\tilde{h}/d\xi^4) + \tilde{h} = \begin{cases} 1 & \text{for } -\tilde{L}_1 \leq \xi \leq 0 \\ 1 + \beta\tilde{p} & \text{for } 0 \leq \xi \leq 1 - \epsilon \\ 1 + \beta(\tilde{p} - \tilde{p}_0) & \text{for } 1 - \epsilon \leq \xi \leq 1 \end{cases} \quad (2-6)$$

The solution algorithm thus consists of determining values for \tilde{h}_2 and $\Delta\tilde{h}$ by the use of the secant method. Equation (2-6) is solved subject to the previously stated elasticity boundary conditions. Convergence is achieved

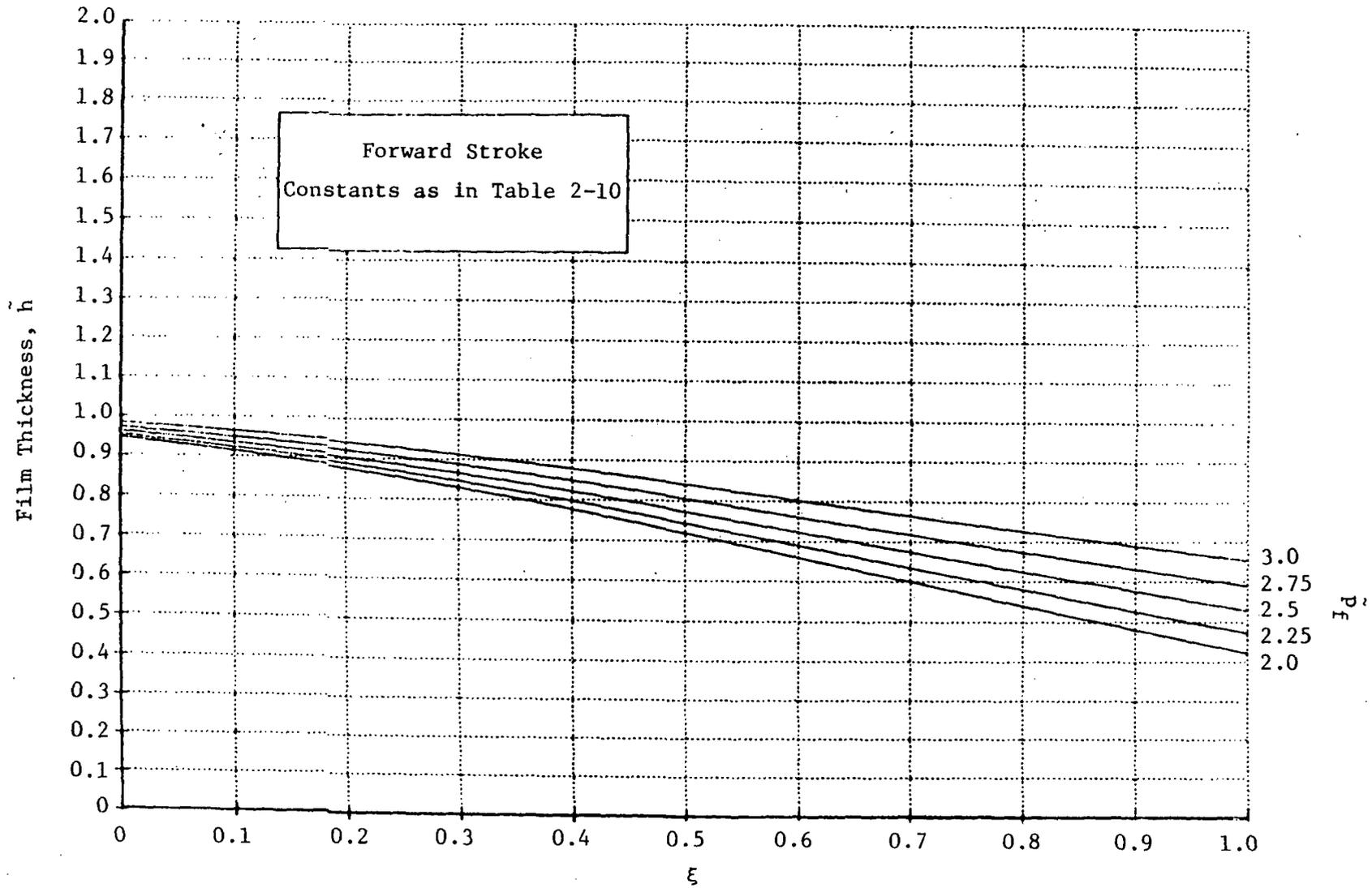


Fig. 2-3 Shape of Film Thickness for Various Values of \tilde{p}_f

when the values of $\tilde{h}(1)$ and $\tilde{h}'(1)$ computed from the solution to Equation (2-6) are within the prescribed tolerance limit of the values of \tilde{h}_2 and $-\Delta\tilde{h}$ used in calculating \tilde{p} from Equations (2-4) and (2-5).

2.3 The Backstroke

2.3.1 Analytical Approach

The previous analysis assumed clamping during the backstroke. This is valid for very high loadings, or for highly flexible pumping rings. Lower ring loadings which do not cause clamping during the entire backstroke may be desirable for pumping ring design in order to reduce wear. Also, when the upstream pressure is high, the ring may stay open during the reverse stroke, even under high clamping forces. Thus, backstroke effects are here added on to the analytical model.

The basic equations remain unchanged except for two important aspects. One is that the shaft motion will be in the negative x direction; thus Reynolds Equation assumes the form

$$\tilde{dp}/d\xi = -[(\tilde{h} - K)/\tilde{h}^3] \quad (2-7)$$

The other critical modification consists in the appearance of cavitation. As shown in Figure 2-4, due to the divergence of the film in the direction of motion, there may not be enough fluid to fill the gap at sufficiently high values of h . The fluid film will then break up into a pattern of streamers similar to that which occurs in the diverging films of a journal bearing. From continuity requirements, the downstream boundary conditions of ξ_c , where the film ends, must then satisfy the following boundary conditions:

$$\begin{aligned} (d\tilde{p}/d\xi) \text{ at } \xi = \xi_c &= 0 \\ \tilde{p}(\xi_c) &= 0 \end{aligned} \quad (2-8)$$

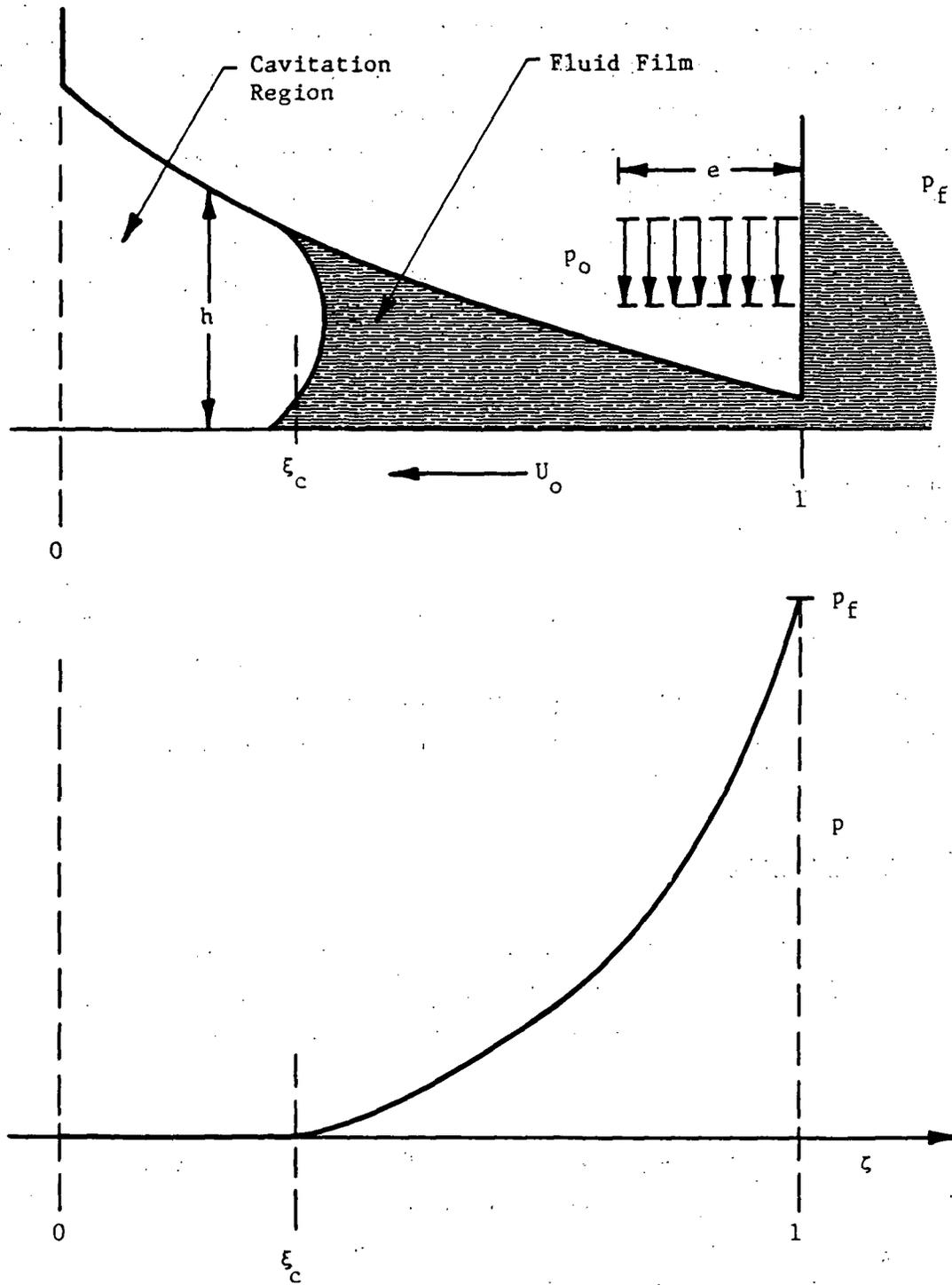


Fig. 2-4 Pumping Ring During Reverse Stroke

To account for cavitation during the backstroke, the equations are integrated backward from $\xi = 1$. As for the forward stroke solution, values of \tilde{h}_2 and $\Delta\tilde{h}$ must be determined by using the secant method. Selected values for \tilde{h}_2 and $\Delta\tilde{h}$ are used in determining $\tilde{h}(\xi)$ from Equation (2-4) which is in turn substituted in the Reynolds Equation (2-7). The resulting equation is then integrated analytically, subject to the constraints that $\tilde{p}(1) = \tilde{p}_f$, and either the condition given by Equation (2-8) for $0 < \xi_c < 1$ or the condition $\tilde{p}(0) = 0$ if cavitation is not predicted to occur. These conditions are sufficient to determine the constant, K , ξ_c (if applicable), and the pressure distribution, $\tilde{p}(\xi)$. Equation (2-6) may now be integrated backward from $\xi = 1$, with the boundary condition at $\xi = 1$ such that

$$\begin{aligned}\tilde{h}(1) &= \tilde{h}_2 \\ \tilde{h}'(1) &= -\Delta\tilde{h} \\ \tilde{h}''(1) &= \tilde{h}'''(1) = 0\end{aligned}$$

The secant method is then used to find the values of \tilde{h}_2 and $\Delta\tilde{h}$ that make the quantities $|\tilde{h}(-\tilde{L}_1) - 1|$ and $|\tilde{h}'(-\tilde{L}_1)|$ be within prescribed tolerance limits. The solutions so obtained provide dimensionless values of the film thickness profile, the pressure profiles and the flow rates for prescribed values of parameters α , β , \tilde{p}_0 , \tilde{p}_f , ϵ , and \tilde{L}_1 .

2.3.2 Nature of Solution

The nature of the solution for the backstroke depends very much on the value of p_f relative to the clamping force, p_0 . It is thus first necessary to describe the conditions which are apt to generate either high or low levels of p_f .

If a pumping ring delivers oil to a closed reservoir of finite volume, the parameters α , β , ϵ , and \tilde{p}_0 describing the pumping ring will predetermine the maximum p_f that the pumping ring is likely to generate. At that point, i.e., when p_{fm} is reached, there will be no further inflow into the reservoir, and the pressure in the reservoir will be maintained by pumping ring action at p_{fm} . In Figure 2-5 this situation would be represented by both valves A and B being closed. However, should there be an outflow from the reservoir, that

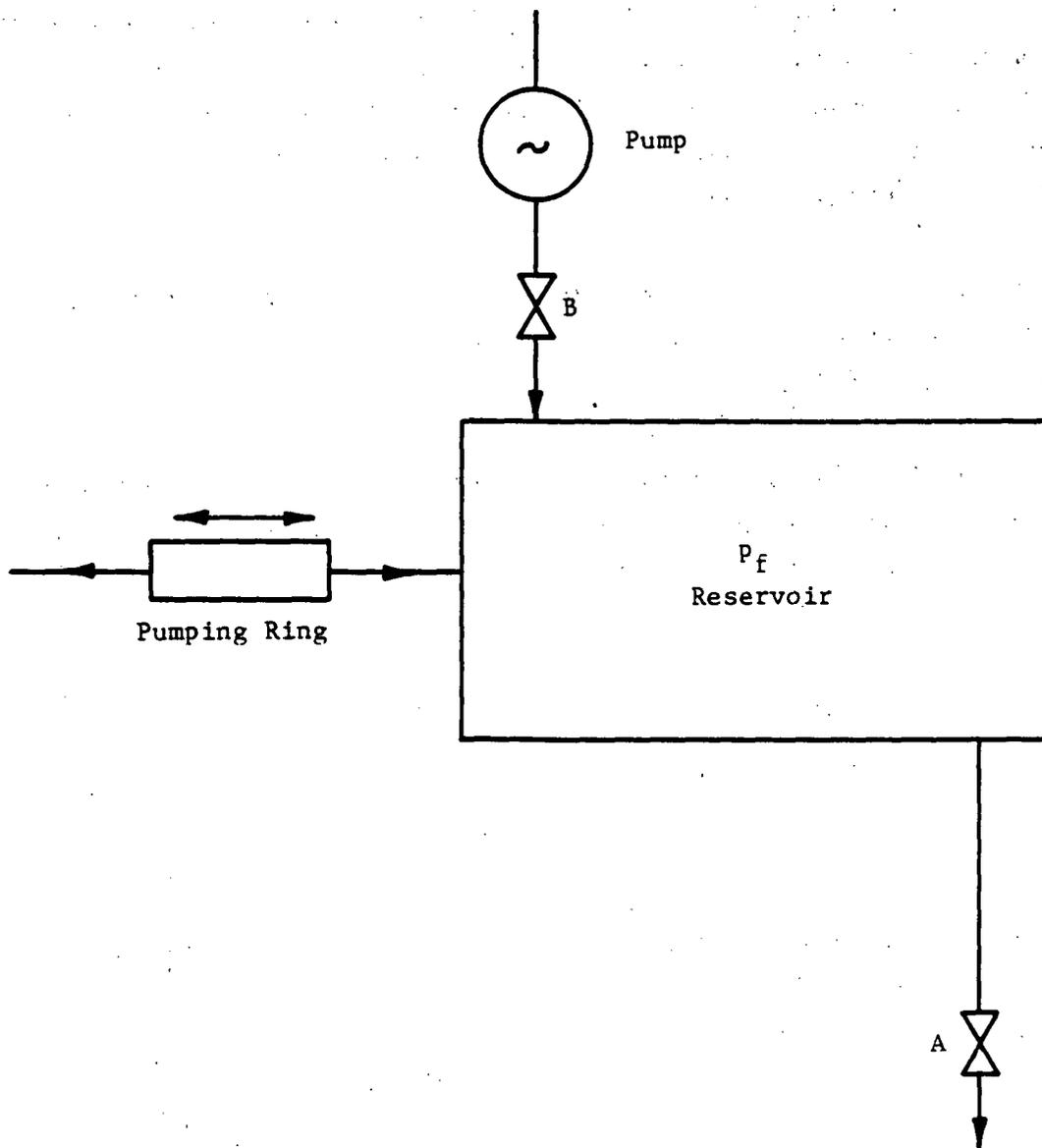


Fig. 2-5 Conditions Determining Level of P_f

is, were valve A open, then the pumping ring would be delivering a net inflow equal to the amount of outflow, with the pressure in the reservoir below the maximum, i.e., $p_f < p_{fm}$. On the other hand, if an external pump were to deliver a certain flow, then it is possible to have $p_f > p_{fm}$. The pumping ring would then be overloaded. The net flow over a cycle would be negative and the outflow through the pumping ring would equal the inflow delivered by the pump. Since there is no external pump in the present system, the overloaded condition will be of academic interest only.

The interaction of pumping ring behavior versus conditions in the reservoir has a direct bearing on the nature of the backstroke solution. The mechanism can be summarized as follows:

- If p_f is low, i.e., if there is an appreciable outflow from the reservoir, the film during the backstroke always cavitates. It is only by having a cavitating backstroke that a net inflow can be maintained over a complete cycle.
- At high values of p_f , there are generally two possible solutions, namely:
 - High h_2 with a noncavitating film
 - Low h_2 with a cavitating film

Such a double set of possible solutions is shown in Figure 2-6, both the same $\tilde{p}_f = 2.7$ and $\tilde{p}_o = 3.5$. The performance of the pumping ring under these two sets of conditions is as follows:

	Noncavitating	Cavitating
Back \tilde{h}_2	0.482	0.120
Forward \tilde{h}_2	0.580	0.580
Back K	-1.433	-0.225
Forward K	-0.42	-0.42
Net K	-1.85	-0.642
ξ_{cav}	--	0.87

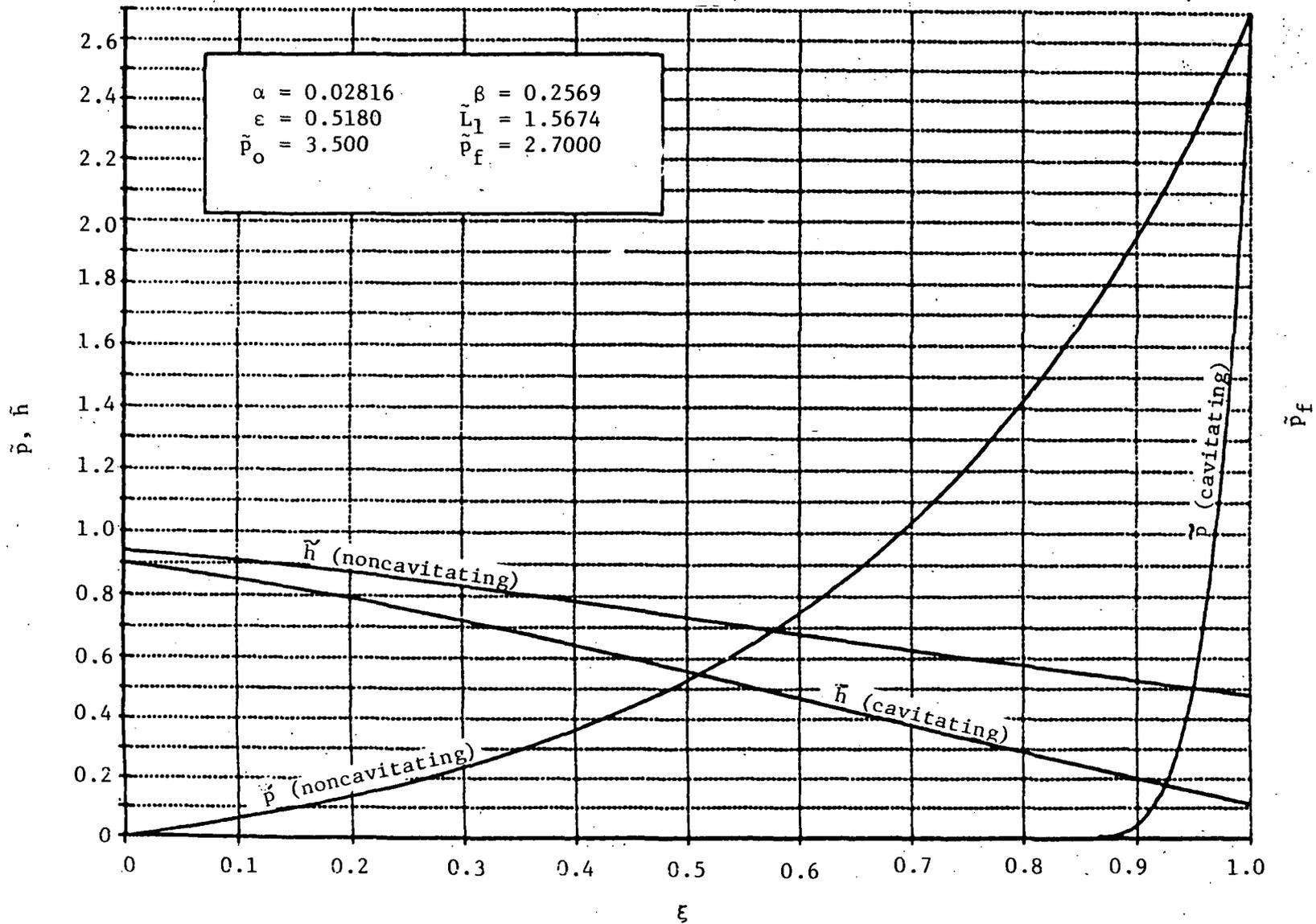


Fig. 2-6 Two Possible Reverse Stroke Solutions for High Values of \tilde{p}_f

- If there is no supply of oil to the reservoir by a pump, that is, if the pressure, p_f , is generated by the pumping ring only, then the solution of a high h_2 with a noncavitating film is practically impossible, since a thick noncavitating film would produce a net outflow so that no high p_f producing that outflow could possibly be generated in the first place.

Thus, in conclusion, for practical cases of an active pumping ring, the backstroke is always accompanied by cavitation.

2.3.3 Performance Characteristics

For the practical range of ring operation, the pumping ring will, in most cases, cavitate during the backstroke. When the condition of $(\tilde{d}p/\tilde{d}\xi)|_{\xi_c} = 0$ is introduced during the backstroke, the resulting film thickness, pressures, and flows are significantly altered from the full film case. Table 2-1 gives the detailed characteristics of cavitating pumping rings as a function of sealed pressure, \tilde{p}_f . Particularly significant for these purposes is the amount of back flow and its effect on the net pumping accomplished over a complete cycle. Figures 2-7 and 2-8 show the pressure distribution during the forward and reverse strokes for a range of \tilde{p}_f from 0 to 5. As seen, cases of $0 \leq \tilde{p}_f \leq 3$ produce cavitation, and the fluid film and corresponding positive pressures for these cases occupy merely a fraction of the interspace. This reduced pressure profile naturally has a strong effect on the film thickness distribution shown in Figures 2-9 and 2-10. Although \tilde{h}_2 for the noncavitating cases, $\tilde{p}_f > 3$, is about the same for the forward and backstrokes, there is a manifold (5 to 10 times) decrease in \tilde{h}_2 during a cavitating backstroke. The shapes of the pressure profiles for a range of values of \tilde{p}_0 (all previous plots were for $\tilde{p}_0 = 3.0$) are shown in Figure 2-11.

A scrutiny of the data contained in Table 2-1, as well as of the accompanying plots, reveals the following interesting features regarding cavitating versus noncavitating films.

TABLE 2-1

PERFORMANCE OF CAVITATING PUMPING RINGS

$\alpha = 0.0282$

$\beta = 0.257$

$\epsilon = 0.518$

$\tilde{p} = 3.5$

$\tilde{L}_1 = 1.57$

\tilde{p}_f	Flow, K			ξ_c	\tilde{h}_{min}		(p_{max}/p_f)
	Forward Stroke	Reverse Stroke	Net		Forward Stroke	Reverse Stroke	
0	0.355	0.0452	0.310	1.0	0.213	0.045	0.74
1.0	0.315	0.0635	0.252	0.983	0.285	0.045	1.011
1.5	0.244	0.0759	0.168	0.975	0.342	0.052	1.0
2.0	0.0877	0.0954	-0.008	0.963	0.422	0.059	1.0
2.5	-0.227	0.903	-1.130	0.0604	0.527	0.342	1.0
3.0	-0.786	2.068	-2.854	None	0.652	0.603	1.0
3.5	-1.6650	3.284	-4.949	None	0.790	0.771	1.0

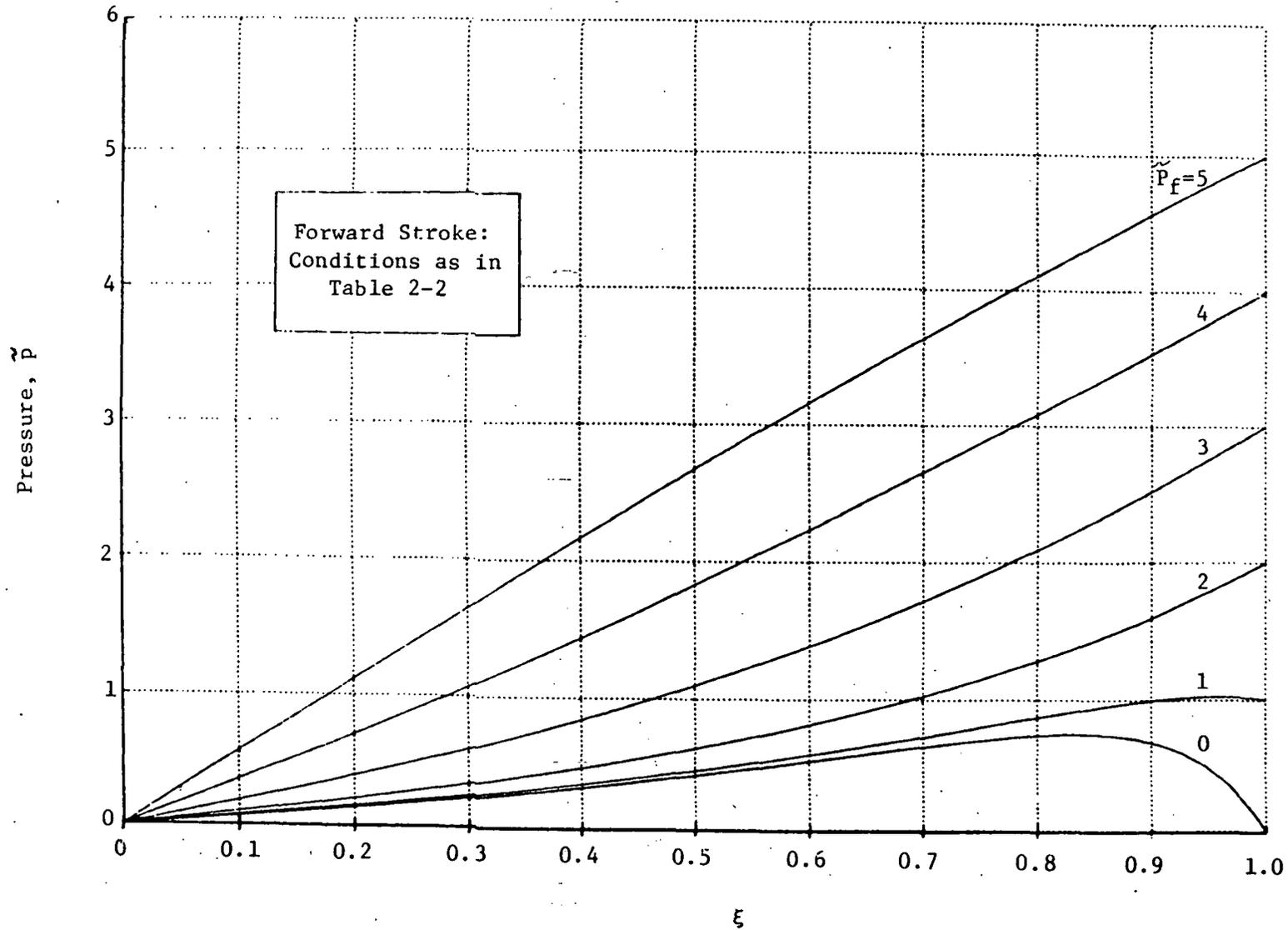


Fig. 2-7 Pressure Distribution During Forward Stroke

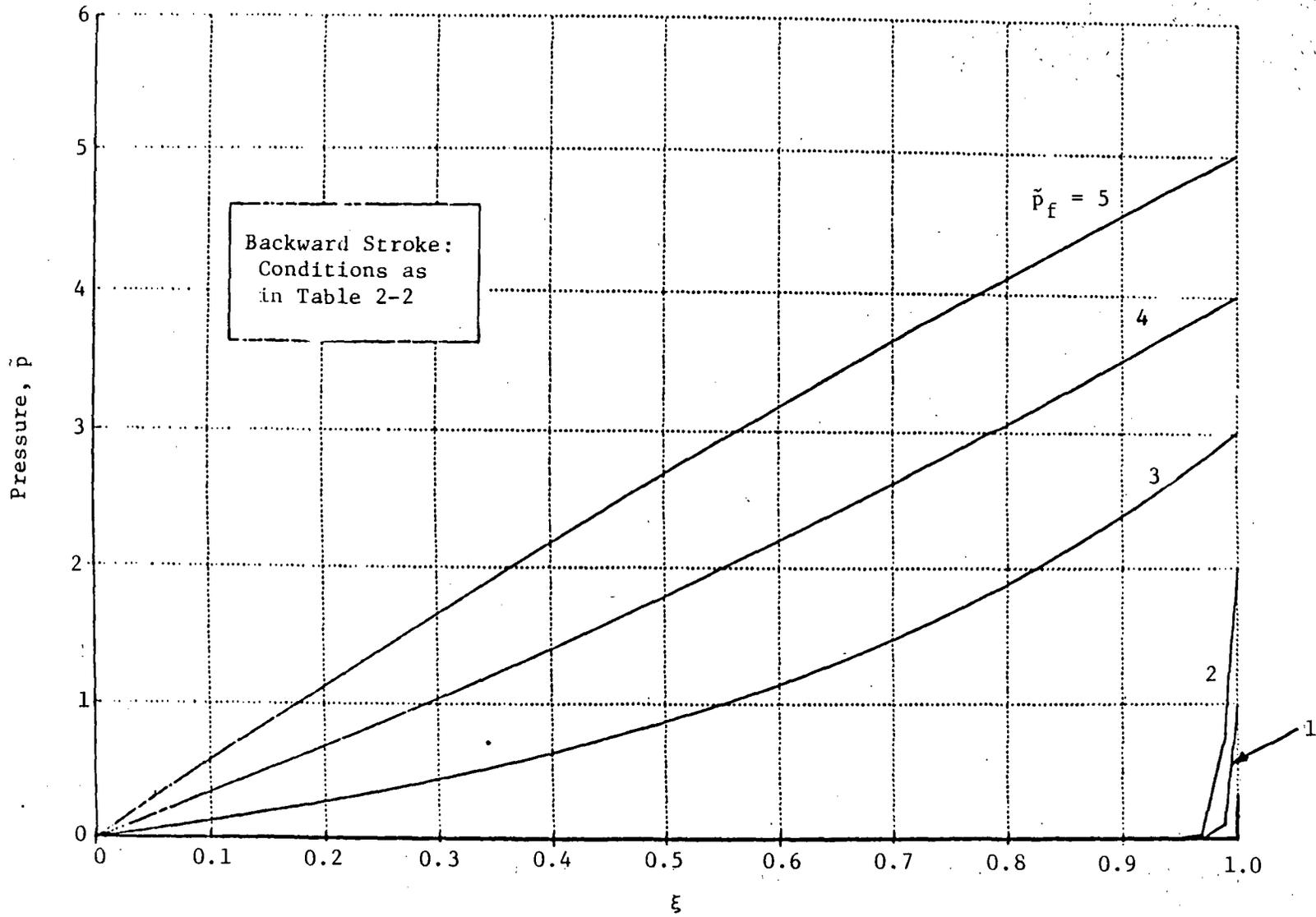


Fig. 2-8 Pressure Distribution During Reverse Stroke

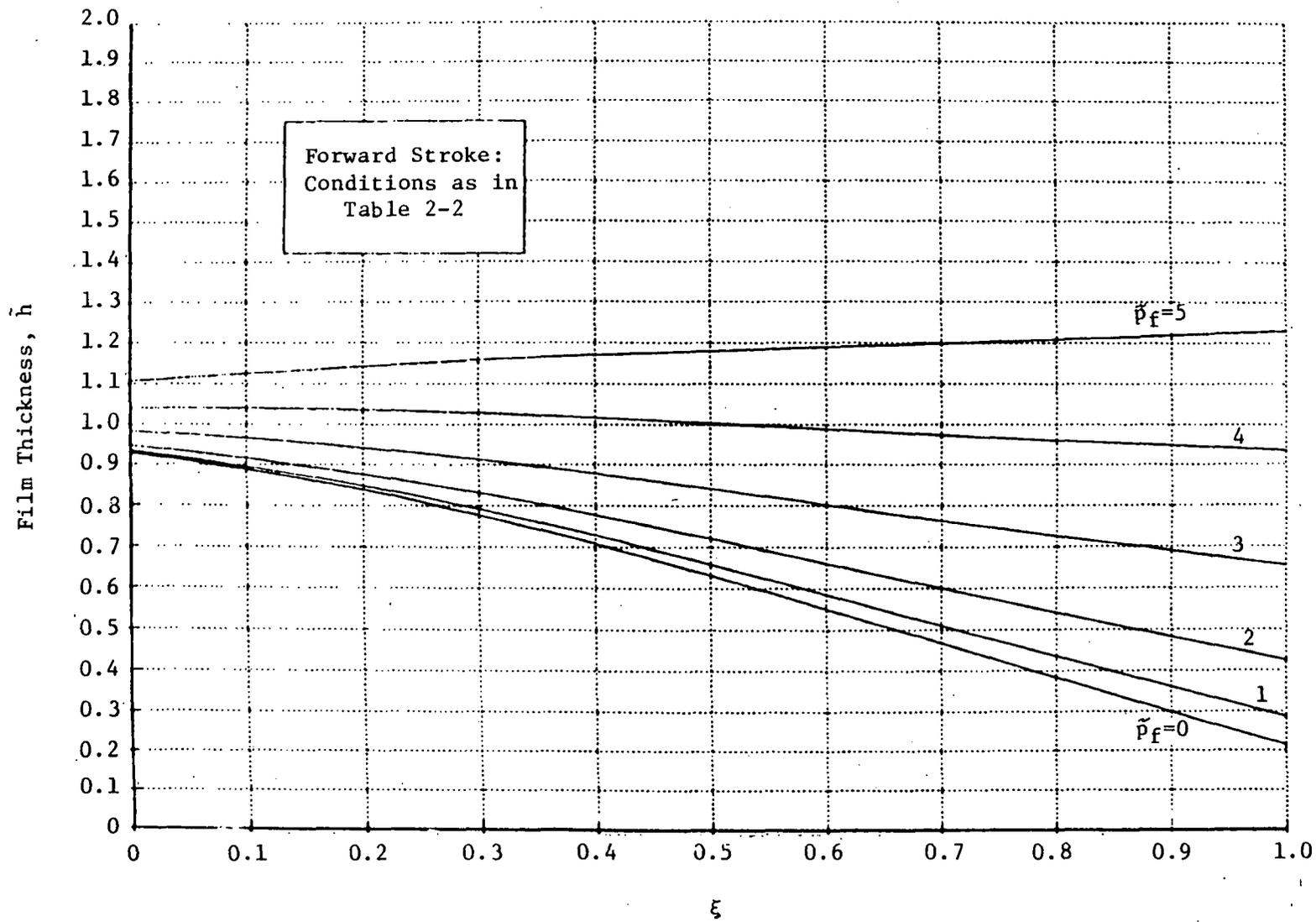


Fig. 2-9 Film Thickness Distribution during Forward Stroke

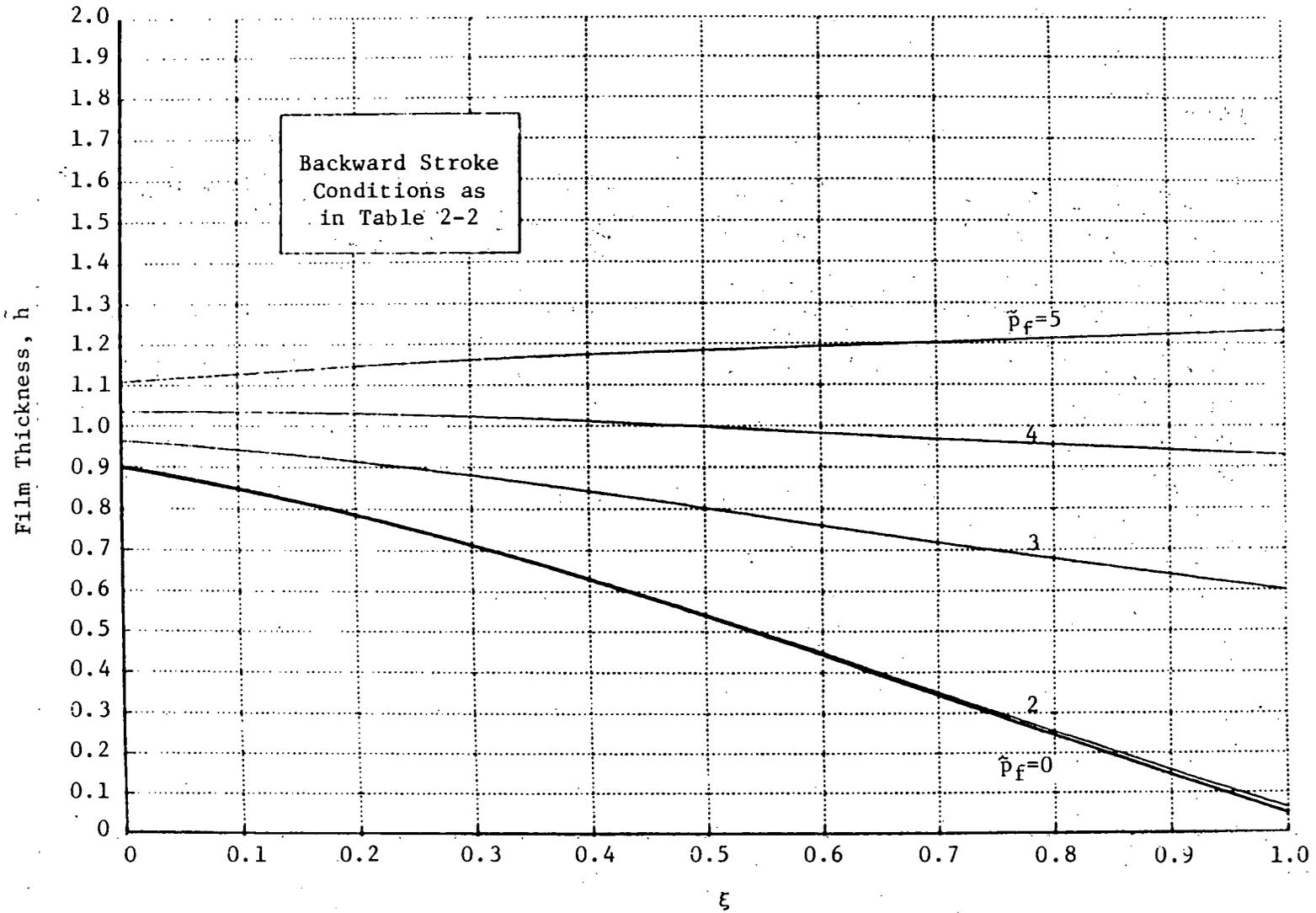


Fig. 2-10 Film Thickness Distribution during Reverse Stroke

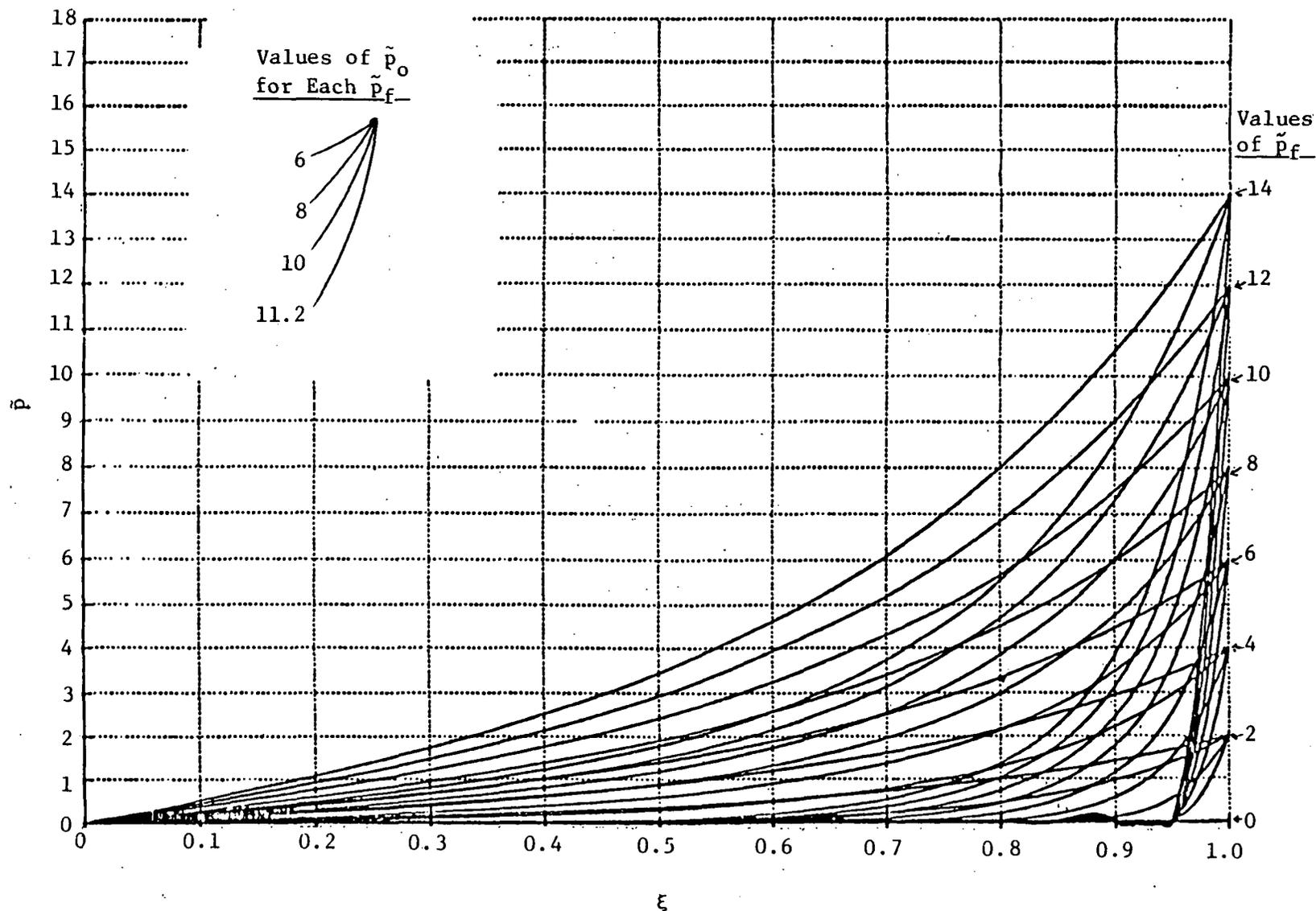
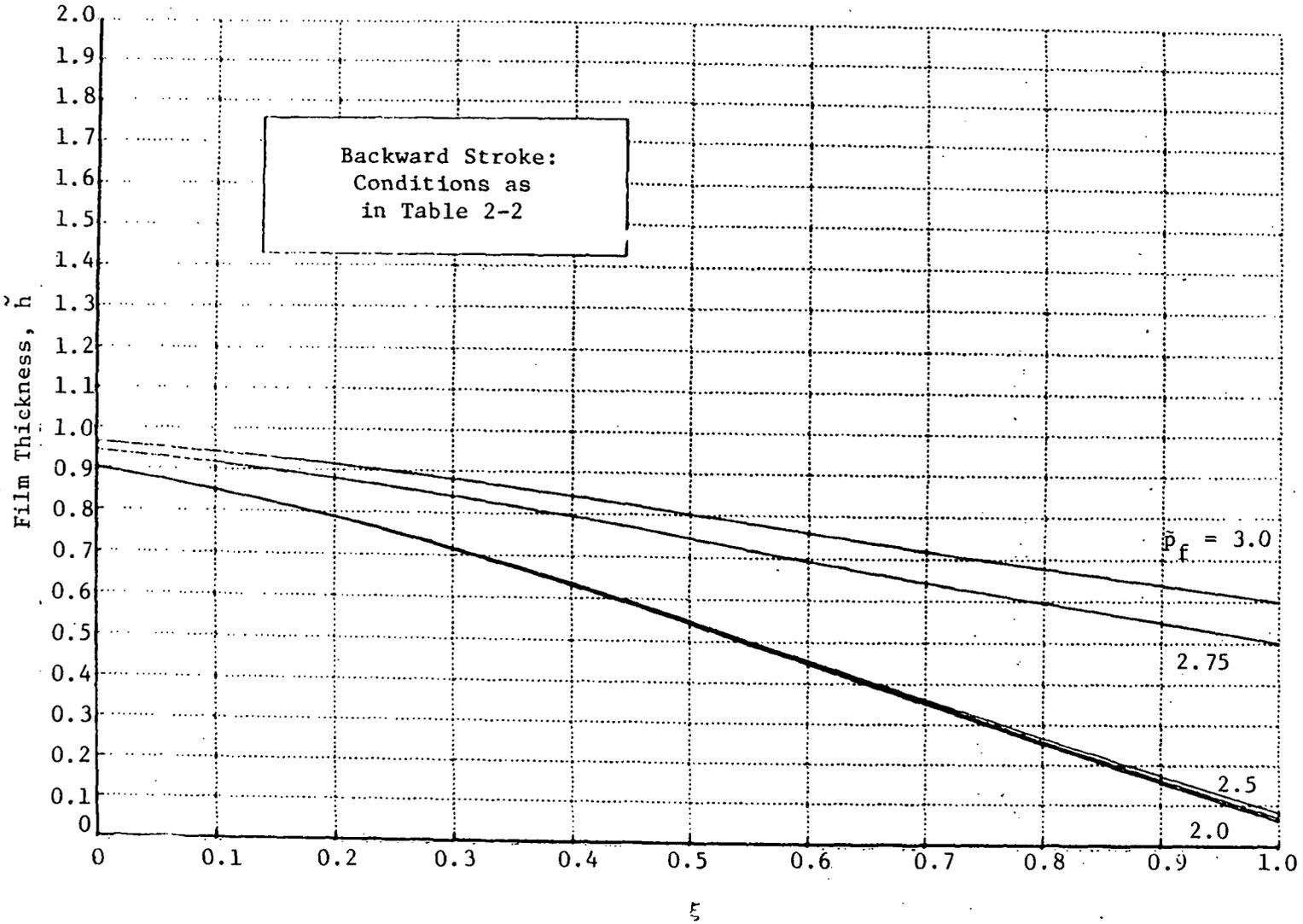


Fig. 2-11 Pressure Profile for Reverse Stroke

ORIGINAL PAGE IS
OF POOR QUALITY

- The change from a cavitating to a noncavitating film is rapid. This is shown in Figures 2-12 and 2-13 where the transition region ($2.0 < \tilde{p}_f < 3.0$) is plotted in some detail. The onset of cavitation seems to occur at about $\tilde{p}_f = 2.5$ where $\xi_c \approx 0$.
- Noncavitating films seem to be accompanied by a net reverse flow even during the forward stroke, emphasizing the previous remark that such cases would be of little practical interest in the application of pumping rings.
- The total net flow, including both the forward and backstrokes, becomes negative, even for cavitating films at about $\tilde{p}_f \approx 2.0$.



ORIGINAL PAGE IS
OF POOR
QUALITY

Fig. 2-12 Film Thickness at Transition from Cavitating to Noncavitating Condition

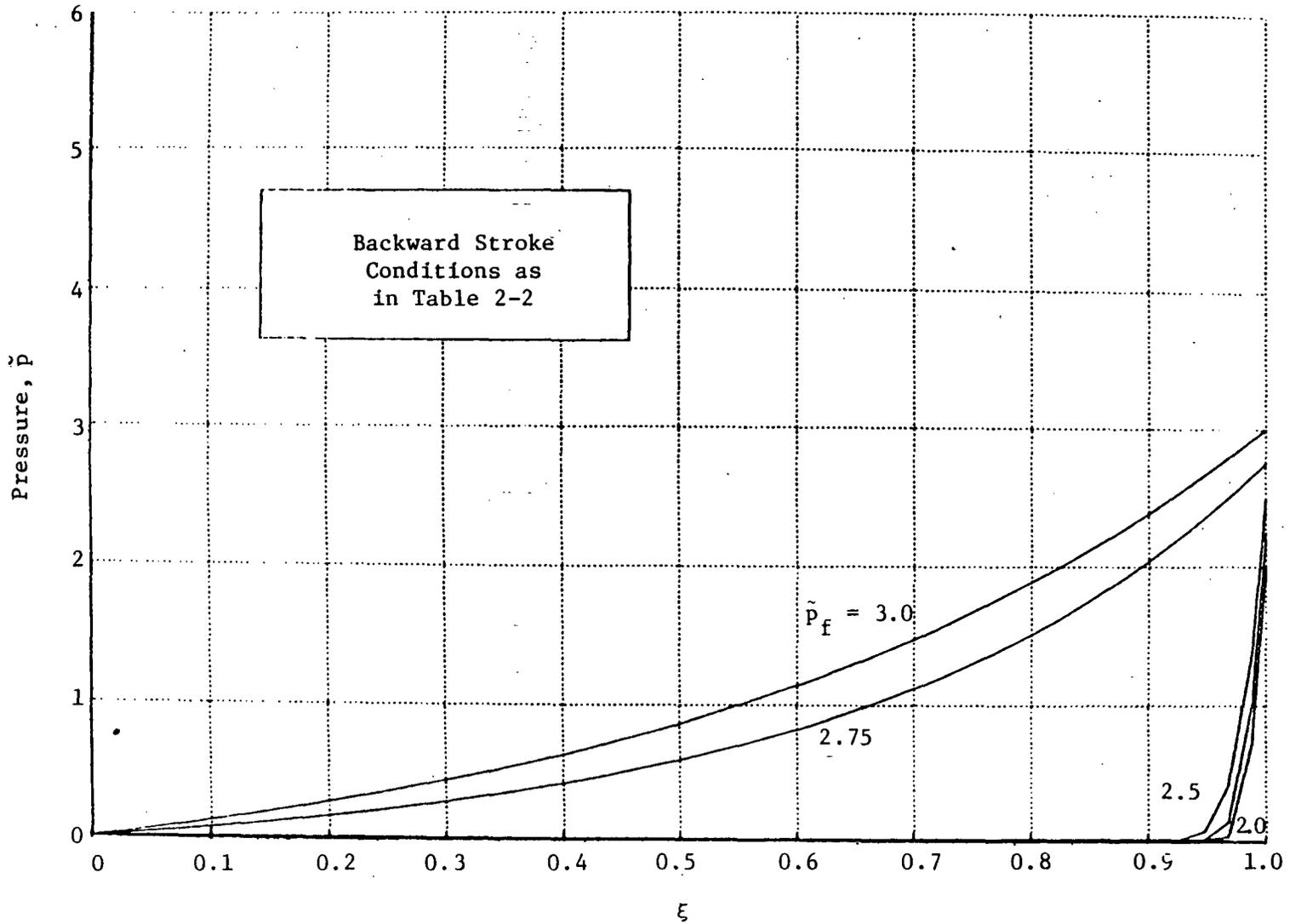


Fig. 2-13 Pressure Profile at Transition from Cavitating to Noncavitating Condition

3.0 ANALYSIS WITH VARIABLE PARAMETERS

This part of the analysis not only abandons the assumptions of constant speed and constant viscosity but also considers some other factors which were previously neglected. Specifically, the analysis included the following elements:

- Thermal effects
- Variable speed
- Squeeze film action
- Starvation
- Nonparallel contours.

Of course, the analysis also retained the backstroke and accompanying cavitation. However, since for the cavitating backstroke, the film thickness is of limited extent, the refinements of variable temperature were not included for that part of the cycle. Thermal effects and transient effects were considered separately to obtain their individual influence. This provides the quantitative corrections associated with these effects without adding the high degree of complexity of a fully coupled analysis.

3.1 Thermal Effects

3.1.1 The Energy Equation

Since no side leakage exists, the one-dimensional energy equation could be used. This, of course, assumes that temperature variation with y can be averaged, a fact which, as will be seen later, is particularly relevant here. It was also assumed that all the heat generated is convected away by the lubricant.

Ignoring conduction, the one-dimensional energy equation can be written as,

$$\rho c u (\partial T / \partial x) = \mu [(\partial u / \partial y)^2] \quad (3-1)$$

Since, from one-dimensional bearing theory

$$u = \frac{1}{2\mu} \left(\frac{\partial p}{\partial x} \right) y (y - h) + U_0 \left(\frac{h - y}{y} \right) \quad (3-2)$$

Equation (3-1) can be integrated with respect to y in the interval $0 \leq y \leq h$ to yield

$$\rho c \left(\frac{U_o h}{2} - \frac{h^3}{12\mu} \frac{\partial p}{\partial x} \right) \left(\frac{\partial T}{\partial x} \right) = \left[\frac{h^3}{12\mu} \left(\frac{\partial p}{\partial x} \right)^2 + \frac{\mu U_o}{h} \right] \quad (3-3)$$

or

$$\left(\frac{\partial T}{\partial x} \right) = \frac{1}{\rho c} \left[\frac{\frac{h^3}{12\mu} \left(\frac{\partial p}{\partial x} \right)^2 + \frac{\mu U_o}{h}}{\frac{U_o h}{2} - \frac{h^3}{12\mu} \frac{\partial p}{\partial x}} \right] \quad (3-4)$$

Normalizing and utilizing the relationship

$$(1/\tilde{\mu})(\partial \tilde{p}/\partial \xi) = (\tilde{h} - K)/(\tilde{h}^3)$$

the following can be obtained

$$\partial T/\partial \xi = [\tilde{\mu}(\xi)/K] \{ [(\tilde{h} - K)^2/(\tilde{h}^3)] + [1/(3\tilde{h})] \} \quad (3-5)$$

where

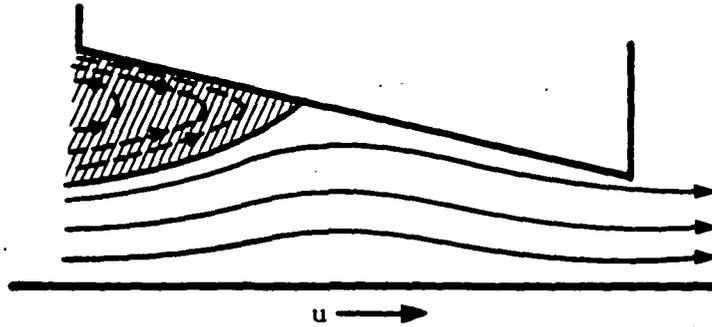
$$\tilde{T} = \frac{6\mu_o U_o L(T - T_o)}{\rho c(C^2)}$$

3.1.2 Lubricant Flow

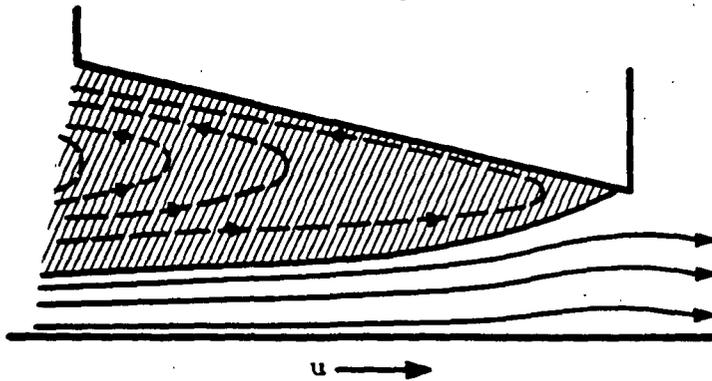
The flow pattern at the interface of a pumping ring was somewhat problematic. The nature of this flow can be visualized in Figure 3-1. As shown in Figure 3-1a, at zero or low upstream pressure, the flow is mainly forward, with only a small pocket of fluid circulating near the inlet of the film. This recirculating flow is induced by the adverse hydrodynamic pressure gradient prevailing near $x = 0$. When the level of p_f rises, more and more of the forward flow near

ORIGINAL PAGE IS
OF POOR QUALITY

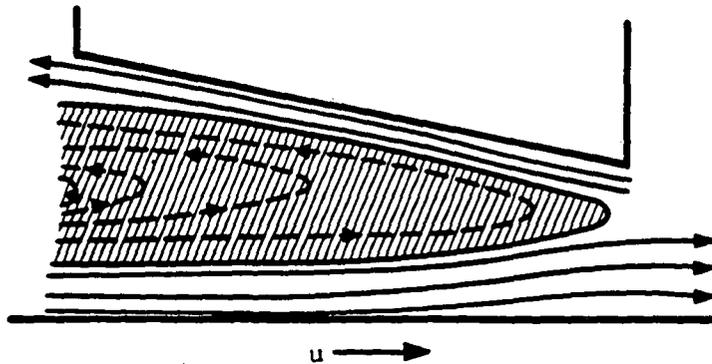
 Recirculating Flow



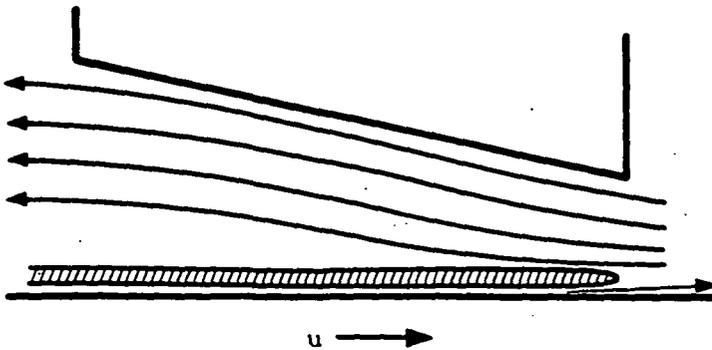
a) $p_f = 0$



b) Moderate p_f



c) High p_f



d) Very High p_f

Fig. 3-1 Lubricant Flow in the Fluid Film

the stationary surface is blocked and eventually reversed (Figure 3-1c) by the static pressure gradient induced by p_f . Eventually, when the upstream pressure is sufficiently high, fluid from the sealed chamber begins to leak backwards along the stationary surface. When the magnitude of p_f becomes very high (Figure 3-1d), most of the flow is backward, with only thin, vanishing layers of forward and recirculating flows maintained near the moving surface.

Thus, in general, there are three layers of flow possessing the following characteristics:

- Forward Flow - This flow is along the moving surface. It enters at a temperature, T_o , that prevails at $x = 0$ and is heated on its way to the reservoir to some T_{max} at $x = L$.
- Recirculating Flow - This flow also enters at a temperature, T_o , but various portions of that flow penetrate only part way into the film before they are reversed and returned to their source. It should be noted that the bulk of the flow recirculates near the entrance, where h is large. It undergoes relatively little viscous shearing, resulting in low energy dissipation to the fluid.
- Reverse Flow - This flow originates in the reservoir entering at a temperature, T_f , and is heated while traveling upstream. Its maximum temperature is reached at the entrance to the pumping ring, at $x = 0$.

In view of these characteristics, only fluid which transverses the whole length, L , i.e., only the forward and reverse flows, was considered instrumental in carrying away the dissipated heat. Since the bulk of the intermediate layer recirculates near the entrance where the temperature differential is relatively small, its effect is left out of the heat balance. This treatment represents a conservative approach because inclusion of the recirculating flow would yield lower temperatures and thus safer operating conditions than those predicted by the present method.

The first task was to find expressions for the three flow regimes in terms of ring geometry and its operating conditions. Defining a dimensionless transverse coordinate and a dimensionless velocity by

$$\tilde{u}(y) = (u/U_0); \quad \eta(\xi) = [y/h]$$

the velocity from Equation (3-2) may be expressed in dimensionless form as

$$u = 3(1 - K/h) \eta(\eta - 1) + (1 - \eta) \quad (3-6)$$

The line of zero velocity, or the line below which all fluid flows forward and above which the flow is backward, is easily obtained from Equation (3-6) by writing $u = 0$ which yields the locus

$$\eta(\xi) \Big|_{u=0} = \eta^*(\xi) = \frac{1}{3[1-K/h(\xi)]} \quad (3-7)$$

At a given ξ , the total flow contained between the moving surface and any point (η, ξ) line is given by

$$\psi(\eta, \xi) = \tilde{h} \int_0^\eta u(\eta') d\eta'$$

and after integration yields

$$\psi(\eta, \xi) = \tilde{h}\eta(\eta - 1)^2 + K\eta^2 (3/2 - \eta) \quad (3-8)$$

By assigning to $\psi(\eta, \xi)$ various constant values, the flow streamlines are obtained. The net flow, of course, is contained between $\eta = 0$ and $\eta = 1$ or

$$\psi(1, \xi) = \tilde{q}_{NET} = (K/2)$$

which is shear flow at $\tilde{h} = K$ where $(d\tilde{p}/d\xi) = 0$.

The film thickness is given by

$$\tilde{h} = \tilde{h}_2 [1 + (a - 1)(1 - \xi)]$$

So that the value of $(K/2)$ for an isoviscous, linear slider is given by

$$\frac{K}{h_2} = \frac{2a}{a+1} (1 - a\tilde{p}_f \tilde{h}_2^2) \quad (3-9)$$

where

$$a = \left(\frac{\tilde{h}_2 - \tilde{h}'_2}{\tilde{h}_2} \right)$$

For the variable viscosity case where $\mu = \mu(\xi)$, the value of K is

$$K = \frac{-\tilde{p}_f + \int_0^1 \frac{\tilde{\mu} d\xi}{\tilde{h}^2}}{\int_0^1 \frac{\tilde{\mu} d\xi}{\tilde{h}^2}} \quad (3-10)$$

The individual streamlines are obtained by assigning different constants to $\psi(\eta, \xi)$ in Equation (3-7). A sample plot of such streamlines for $a = 3$ and different values of $\tilde{p}_f \tilde{h}_2^2$ is shown in Figure 3-2. Reverse flow will commence when the dividing streamline between the forward and recirculating flow reaches $\xi = 1$. This, from Equation (3-7), occurs at

$$\eta^*(1) = \frac{1}{3[1 - K/h_2]} = 1 \quad (3-11)$$

or at $(K/h_2) = 2/3$. At values of $K/h_2^2 < 2/3$, the tangent point of the recirculating envelope at $\xi = 1$ will lie below $\eta^* = 1$, opening up a passage for reverse flow. Below the recirculating envelope the fluid will, as shown in Figure 3-3, flow forward at a rate of

$$\tilde{q}_F = \psi[\eta^*(1), 1]$$

whereas, above the envelope there will be reverse flow at a rate of

$$\tilde{q}_R = \tilde{q}_F - \tilde{q}_{NET} = \psi[\eta^*(1), 1] - (K/2) \quad (3-12)$$

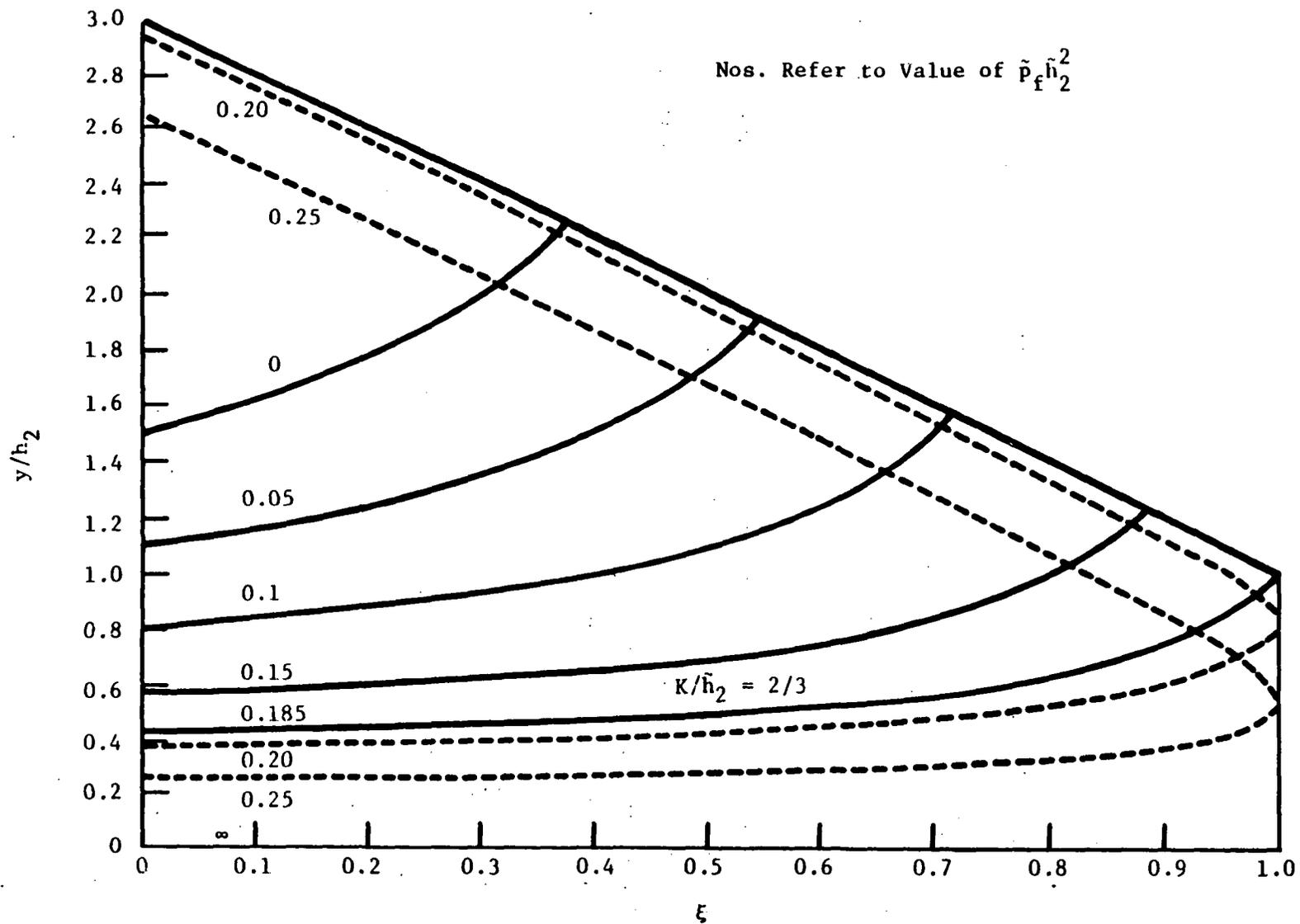


Fig. 3-2 Fluid Film Streamlines for $a = 3$

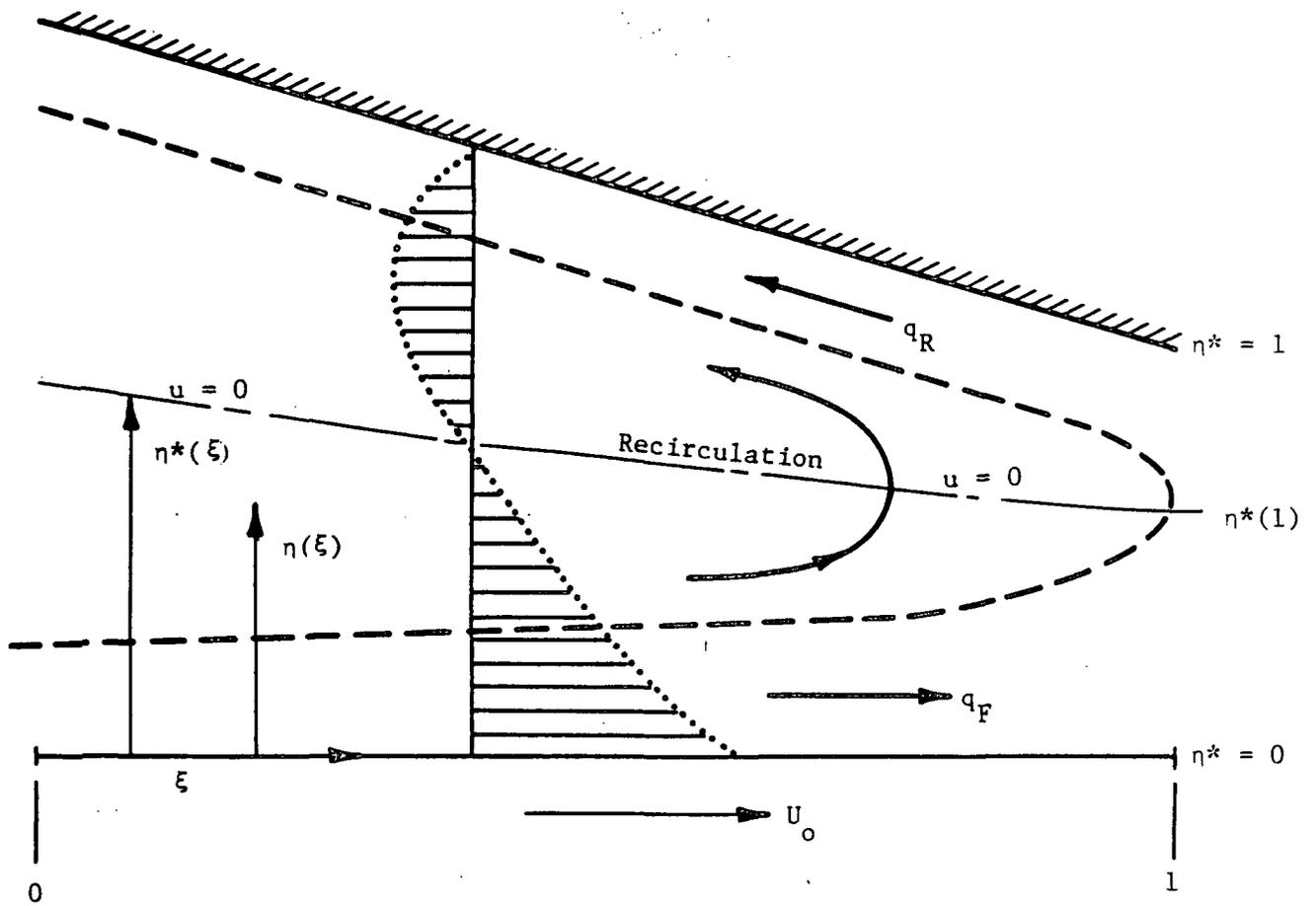


Fig. 3-3 Mapping of the Three Flow Regions

3.1.3 Modeling of Thermal Problem

As stated previously, the viscous heating is assumed to be convected away only by the fluid that passes the interspace, be it to the right or to the left. For this purpose, the interspace is modeled so as to be filled with these two streams, i.e. a forward flow, q_F , and a reverse flow, q_R , as shown in Figure 3-4. By designating

$$f = q_F / (q_F + q_R) \quad (3-13)$$

$f\theta(x)$ will be the heat convected to the right by q_F and $(1-f)\theta$ the heat convected to the left by q_R . Likewise, temperatures will be averaged across the film (i.e., in the y but not in the x direction) with $T_F(x)$ and $T_R(x)$ representing the temperature profiles generated in the two flow layers, q_F and q_R . The overall average temperature profile will be obtained from

$$T(x) = f T_F(x) + (1 - f) T_R(x) \quad (3-14)$$

The viscosity at each x station will be based on this averaged temperature, i.e.

$$\mu(x) = \mu[T(x)].$$

The expression for the viscous heating (Section 3.1.1) is exact in the sense that the losses are calculated over the exact velocity profile, including the region of recirculating flow, namely.

$$\phi(x) = \mu(x) \int_0^h (du/dy)^2 dy \quad (3-15)$$

where $u(x,y)$ is the velocity profile shown in Figure 3-3. Consistent with the flow model, $\mu(x)$ in the calculation of this viscous shear will be that corresponding to the average temperature $T(x)$.

It should be noted that, whereas with no reverse flow ($q_R = 0$), the temperature at the inlet to the film is a constant equal to T_0 , this is no longer true when reverse flow sets in. The reverse flow, starting at an initial temperature, T_f ,

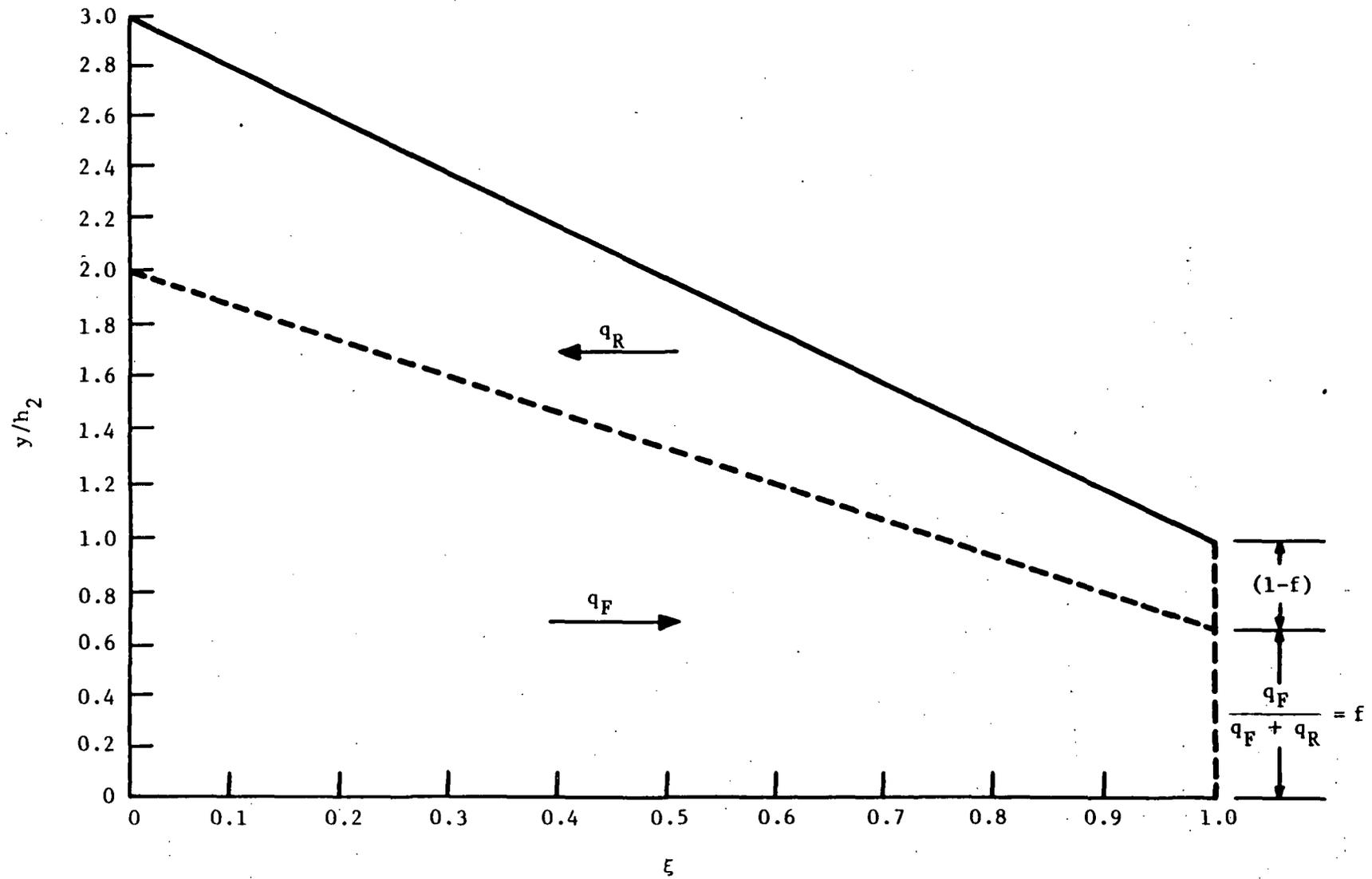


Fig. 3-4 Subdivision of Flow for Convective Heat Transfer

will reach a maximum at $x = 0$ and, since temperatures are averaged across y , the inlet temperature will be some $T_o' > T_o$.

3.1.4 Calculation Procedure

The procedure to be followed in calculating thermal effects in the fluid film is, in view of the previous discussion, as follows:

- a. The differential equation to be solved is

$$\frac{d\bar{p}}{d\xi} = \bar{\mu}(\xi) \left[\frac{\bar{h} - K}{\bar{h}^3} \right]$$

where due to the variable viscosity $\mu(\xi)$

$$K = \frac{\int_0^{\bar{h}} \frac{\bar{\mu}(\xi)}{\bar{h}^2} d\xi - \bar{p}_f}{\int_0^1 \frac{\bar{\mu}(\xi)}{\bar{h}^3} d\xi}$$

and

$$\mu(\xi) = F \left[T_o + \frac{6\mu_o U_o L}{\rho c C^2} \bar{T}(\xi) \right]$$

- b. If $(K/\bar{h}_2) > 2/3$, there is no reverse flow and

$$\frac{d\bar{T}}{d\xi} = \frac{\bar{\mu}(\xi)}{K} \left[\frac{(\bar{h} - K)^2}{\bar{h}^3} + \frac{1}{3\bar{h}} \right] \quad (3-16)$$

with $T = T_o$ at $\xi = 0$.

- c. If $(K/\tilde{h}_2) < 2/3$, there is reverse flow, and the following quantities have to be established in order to account for both the forward and backward streams:

$$\eta^* = \eta^*(1) = \frac{1}{3 \left[1 - \frac{K}{\tilde{h}_2} \right]}$$

$$K_F = 2\tilde{h}_2 \eta^*(\eta^* - 1)^2 + 2K(\eta^*)^2 [3/2 - \eta^*] = 2\tilde{q}_F$$

$$K_R = K_F - K = 2q_R$$

$$K_T = K_F + K_R = 2K_F - K = 2(q_F + q_R) = 2q_T$$

$$\frac{d\tilde{T}}{d\xi} = \frac{\tilde{\mu}(\xi)}{K_T} \left[\frac{(\tilde{h} - K)^2}{\tilde{h}^3} + \frac{1}{3\tilde{h}} \right] \quad (3-17)$$

Equation (3-17) needs a solution with two different boundary conditions:

- For $T_F(\xi)$, $T = T_o$ at $\xi = 0$
- For $T_R(\xi)$, $T = T_f$ at $\xi = 1$

The averaged temperature, any given ξ , is thus

$$\tilde{T}(\xi) + \frac{\tilde{T}_R K_F + \tilde{T}_L K_R}{K_T} \quad (3-18)$$

where

$$\tilde{T} = \frac{\rho c C^2 (T - T_o)}{6\mu_o U_o L}$$

A sample solution for the temperature profile in the fluid film for various values of upstream pressure is given in Figure 3-5. The solutions are for the case of equal upstream and downstream boundary temperatures. Curves which start at $T = 86^\circ\text{F}$ are those without reverse flow and thus have what may be called a conventional profile. However, at $\tilde{p}_f > 1.5$, there is reverse flow and, due to the averaging of temperatures across y , the inlet temperature is higher than

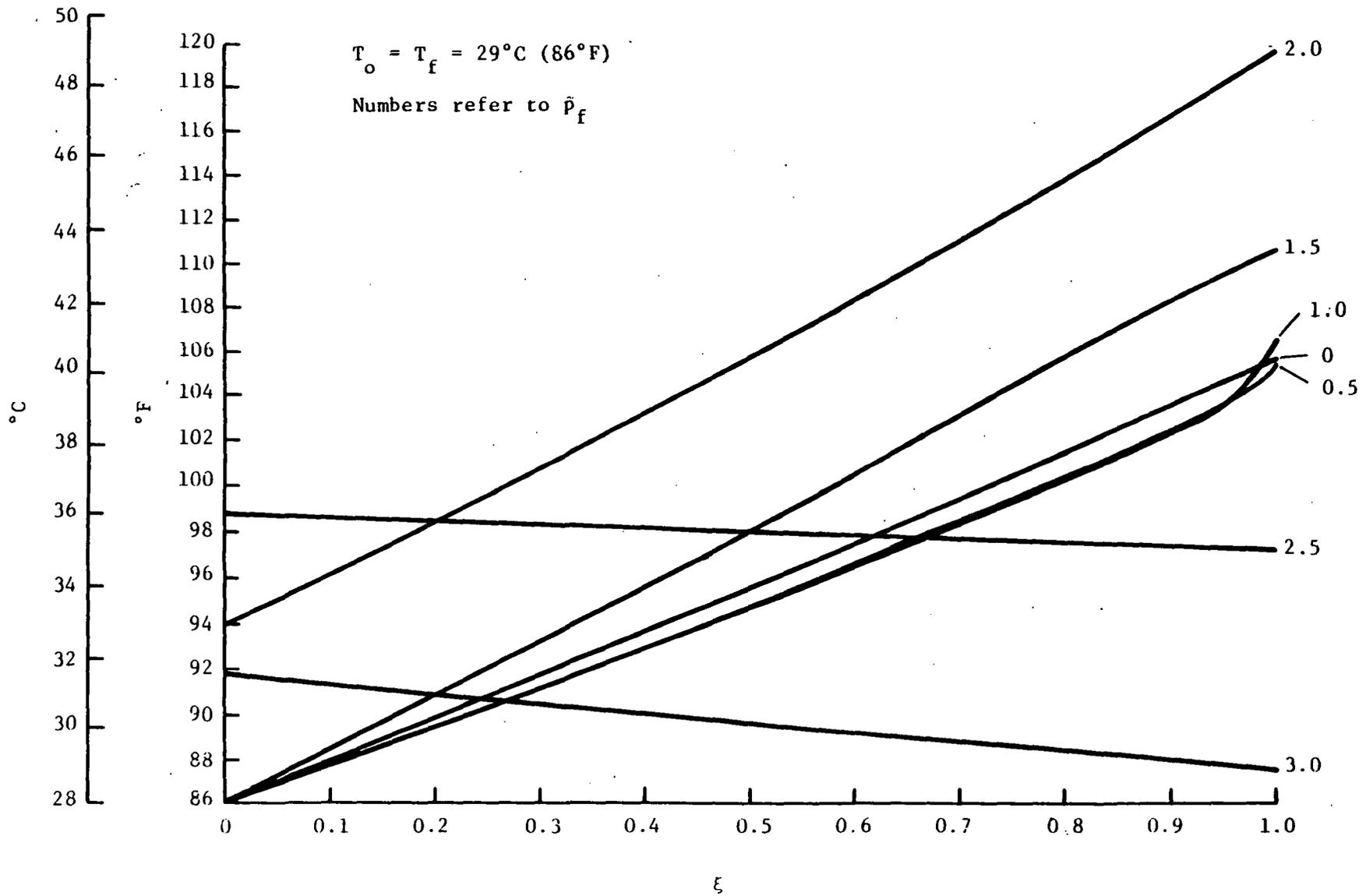


Fig. 3-5 Temperature Profile as Function of \bar{p}_f

86°F. It is interesting to note that, due to the cooling effect of the backward flow, the maximum temperatures in the film actually decrease as the back flow increases above a certain level. Thus, for $p_f = 0$, $T_{\max} = 107^\circ\text{F}$, and for $p_f = 3.0$, $T_{\max} = 92^\circ\text{F}$ and occurs not at the trailing but at the leading edge of the film. The highest temperatures occur at some intermediate combination of forward and backward flows; in this particular example it happens at $p_f = 2.0$ with T_{\max} reaching 120°F .

3.2 Variable Velocity and Squeeze-Film Effects

Throughout the previous discussions, the rod velocity was considered to be constant, given by $U_0 = 2sf$. This, of course, represents the average velocity over each half cycle. In actuality, the rod, driving a crankshaft, moves with a variable velocity given by

$$u = u_{\max} \cos \pi 2ft$$

where

$$u_{\max} = fs = \pi U_0 / 2$$

Consequently, a normal velocity component and squeeze film forces are imposed on the ring, as shown in Figure 3-6. These are generated by a variation of the hydrodynamic forces and of film thickness, as a function of the variable velocity. The normal velocity introduces perturbations on nearly all the relevant quantities, such as film thickness, pressures, flows, extent of cavitation, and others.

When variable velocity and squeeze-film effects are included, the Reynolds Equation becomes:

$$\left(\frac{\partial}{\partial \xi}\right) \left[\tilde{h}^3 \left(\frac{\partial p}{\partial \xi}\right) \right] = (\pi/2) \left[\cos \tau \left(\frac{\partial \tilde{h}}{\partial \xi}\right) + \sigma \left(\frac{\partial \tilde{h}}{\partial \tau}\right) \right] \quad (3-19)$$

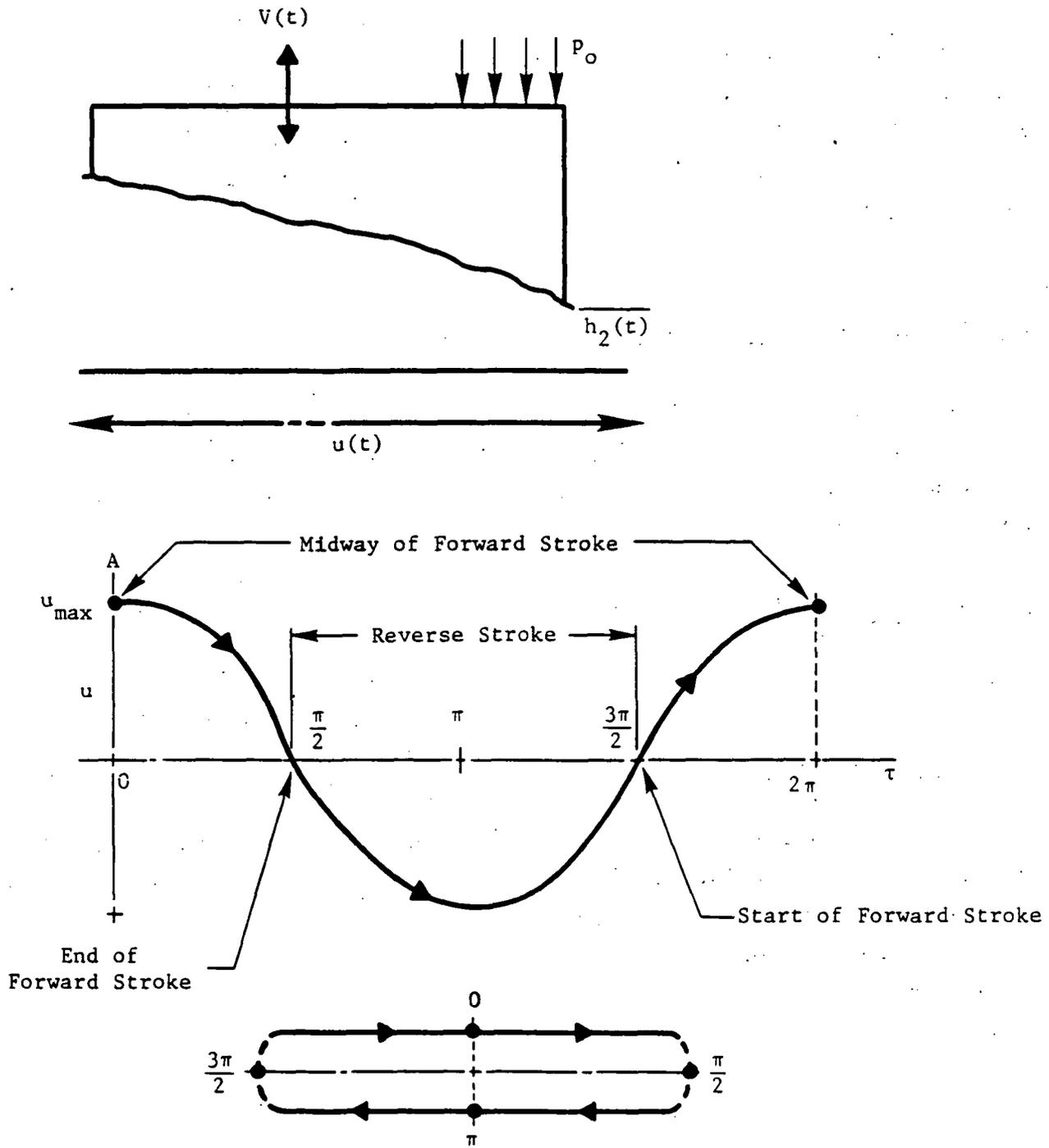


Fig. 3-6 Variable Velocity and Squeeze Film Effects

where σ is the squeeze-film parameter given by $\sigma = 4L/s$, and τ is the dimensionless time $\tau = 2\pi ft$.

The relevant boundary conditions are:

- For no cavitation: $\tilde{p}(0) = 0; \tilde{p}(1) = \tilde{p}_f$
- For cavitation: $\tilde{p}(1) = \tilde{p}_f, \left. \frac{d\tilde{p}}{d\xi} \right|_{\xi = \xi_c} = 0$

The one-dimensional, transient Reynolds Equation, given by Equation (3-19), may be solved with the use of the implicit time-transient method. $\partial h/\partial \tau$ may be written as:

$$(\partial \tilde{h}/\partial \tau) = [\tilde{h}_{(i)} - \tilde{h}_{(i-1)}]/\Delta \tau$$

where the subscript i in parentheses denotes the i^{th} time step, and $\Delta \tau$ is the interval between the i^{th} and $(i-1)^{\text{th}}$ time steps. Hence

$$\tilde{h}_{(i)} = \tilde{h}_{2(i)} + \Delta \tilde{h}_{(i)} (1 - \xi)$$

and the discretized form of Equation (3-19) is written as

$$\frac{d}{d\xi} \left[\tilde{h}_{(i)}^3 \frac{d\tilde{p}_{(i)}}{d\xi} \right] = \frac{\pi}{2} \left\{ [-\cos \tau_{(i)} \Delta \tilde{h}_{(i)}] + \sigma \frac{[\tilde{h}_{(i)} - \tilde{h}_{(i-1)}]}{\Delta \tau} \right\} \quad (3-20)$$

If it is assumed that $\tilde{h}_{(i-1)}$ is known, then Equation (3-20) can be integrated analytically to obtain $\tilde{p}_{(i)}$ as a function of ξ at the i^{th} time step. The elasticity equation, Equation (2-6), remains unchanged except for the time dependence of \tilde{p} , and may be coupled with the solution to Equation (3-20) and solved at each time step as described previously for the steady-state solution. Values of \tilde{h}_2 and $\Delta \tilde{h}$ at the previous time step ($\tilde{h}_{2(i-1)}, \Delta \tilde{h}_{(i-1)}$) were used in evaluating $\tilde{h}_{(i-1)}$ in Equation (3-20) and for initial guesses for $\tilde{h}_{2(i)}$ and $\Delta \tilde{h}_{(i)}$ for use in the secant method.

Quasi-static solutions ($\sigma = 0$) were obtained at the middle of the forward stroke ($\tau = 0$) to start the marching procedure. Solutions were then computed at successive time steps until periodic solutions were obtained.

Sample solutions were run on a pumping ring having the following dimensionless characteristics:

$$\alpha = 0.0282, \beta = 0.2569, \varepsilon = 0.518, \tilde{L}_1 = 1.567, \sigma = 4.0$$

Three cases were considered, at different combinations of \tilde{p}_f and \tilde{p}_o ; the $\tilde{p}_o = 4.0$ solution also represents a case where the ring is clamped on the backstroke. The three cases were all solved for steady-state ($u = u_o$), quasi-static ($\sigma = 0$), and transient conditions to show the effects of squeeze film on ring performance. The value of $\sigma = 4.0$ is quite large for practical applications as it corresponds to a stroke, s , equal to the bearing land, L . For present applications, σ is generally less than 1. The large value was used to exaggerate the effects of squeeze film for purposes of illustration and interpretation. Graphs for film thickness, flow, and pressure profiles are given in Figures 3-7 through 3-14, whereas the values of the component flows are itemized in Table 3-1. Note should be taken that $\tau = 0$ corresponds in these plots to rod position at $u = u_{\max}$, i.e., at the midpoint of its forward stroke.

Considering first the effects of variable velocity alone (quasi-static solution), the plots show the following trends:

- The film thickness variation during the forward stroke follows the sinusoidal shape of the velocity curve. During the backstroke velocity, variation has no effect whenever cavitation commences at the trailing edge and a small effect when cavitation is located at $\xi < 1$ (Figure 3-9).
- The flow curve follows the shape of the velocity curve throughout the cycle, except, of course, when, due to a clamped ring, the flow during the backstroke is zero. In the particular case of Figure 3-14, the flow, after peaking at u_{\max} , became negative at the end of the forward stroke ($u \rightarrow 0$). However, upon commencing the backstroke, the ring clamped and flow ceased.

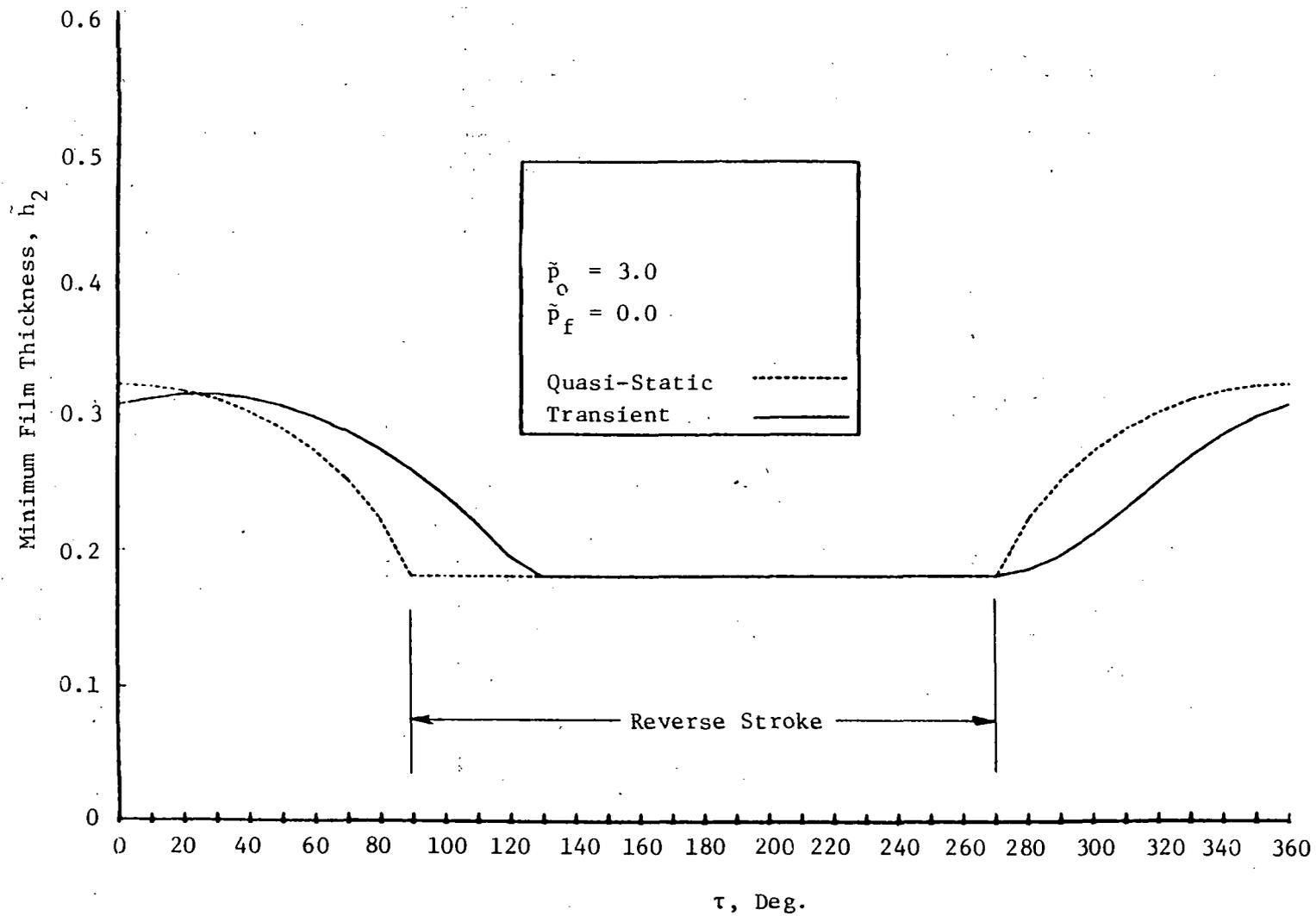


Fig. 3-7 Film Thickness \bar{h}_2 as Function of Time

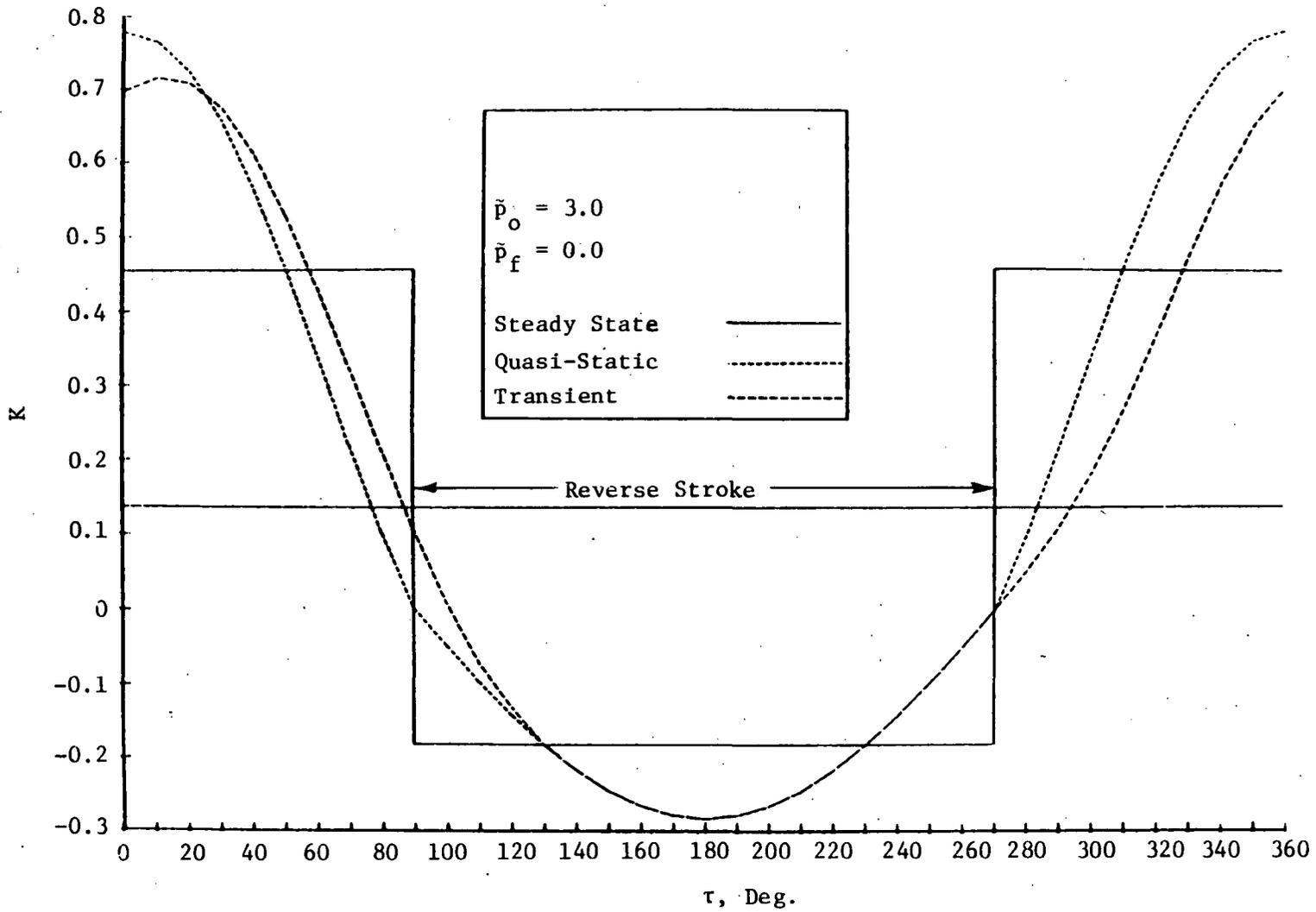


Fig. 3-8 Flow Parameter Versus Time

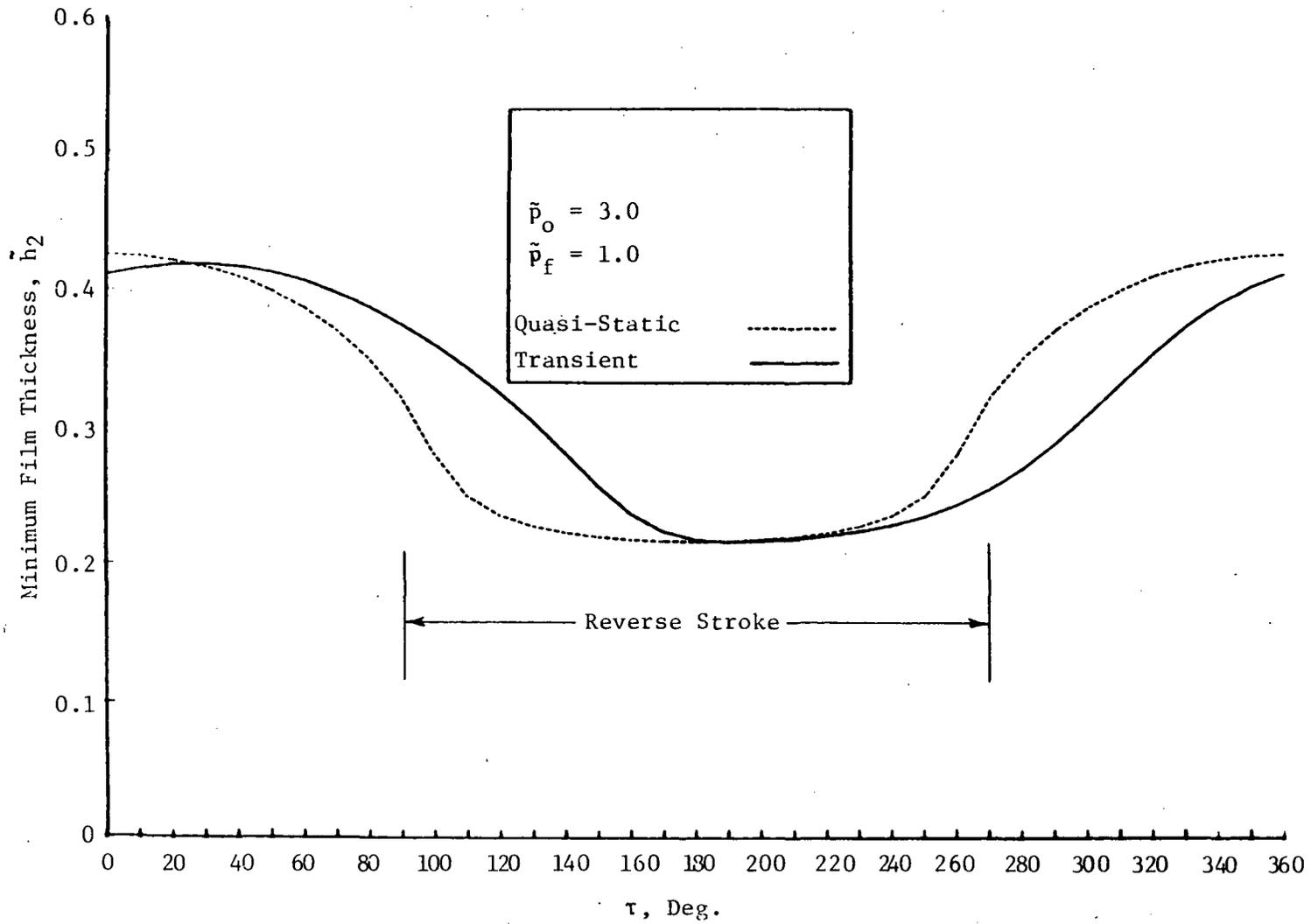


Fig. 3-9 Film Thickness \tilde{h}_2 as Function of Time

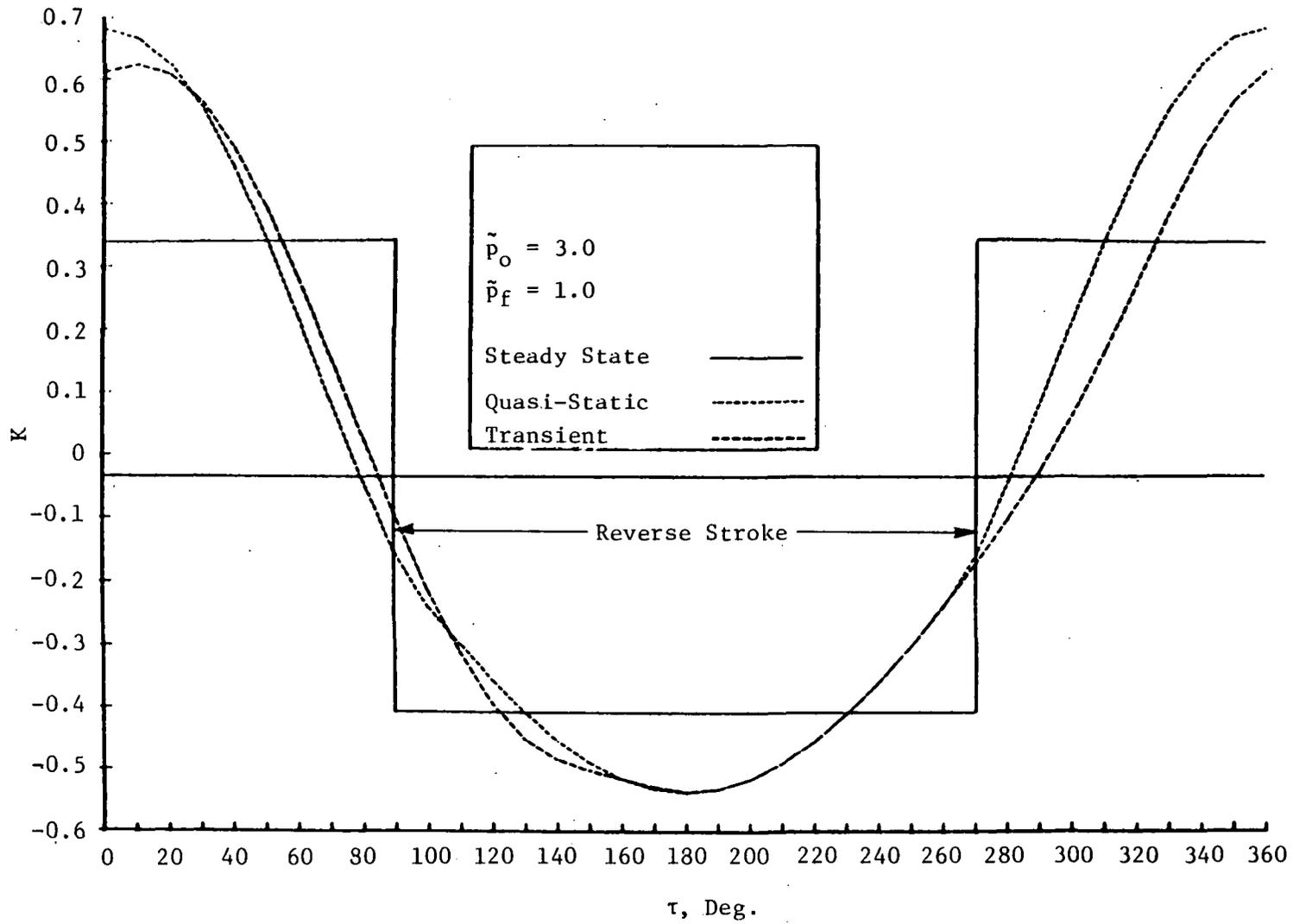


Fig. 3-10 Flow Parameter Versus Time

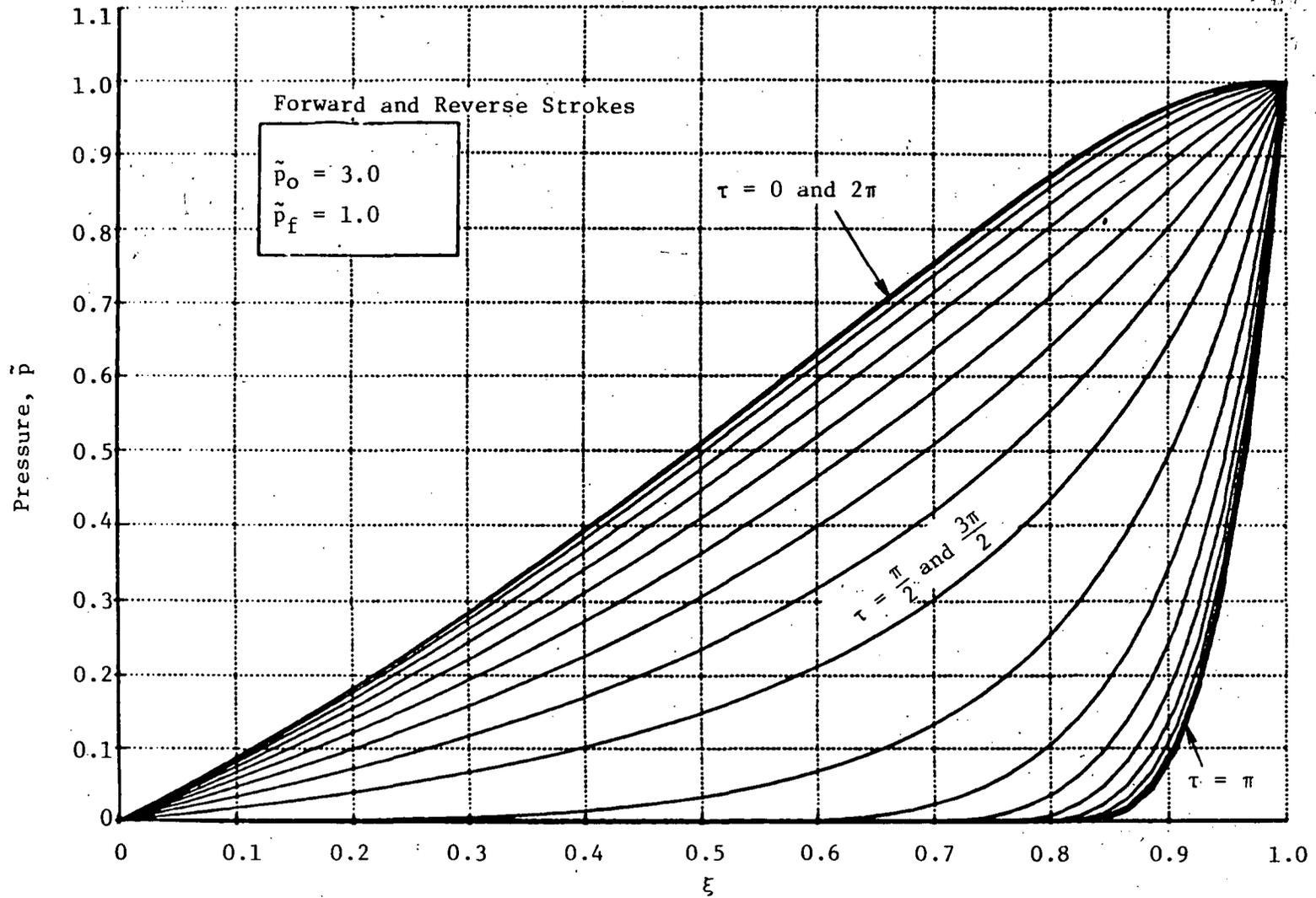


Fig. 3-11 Pressure Profiles as Function of Time for Quasi-Static Analysis

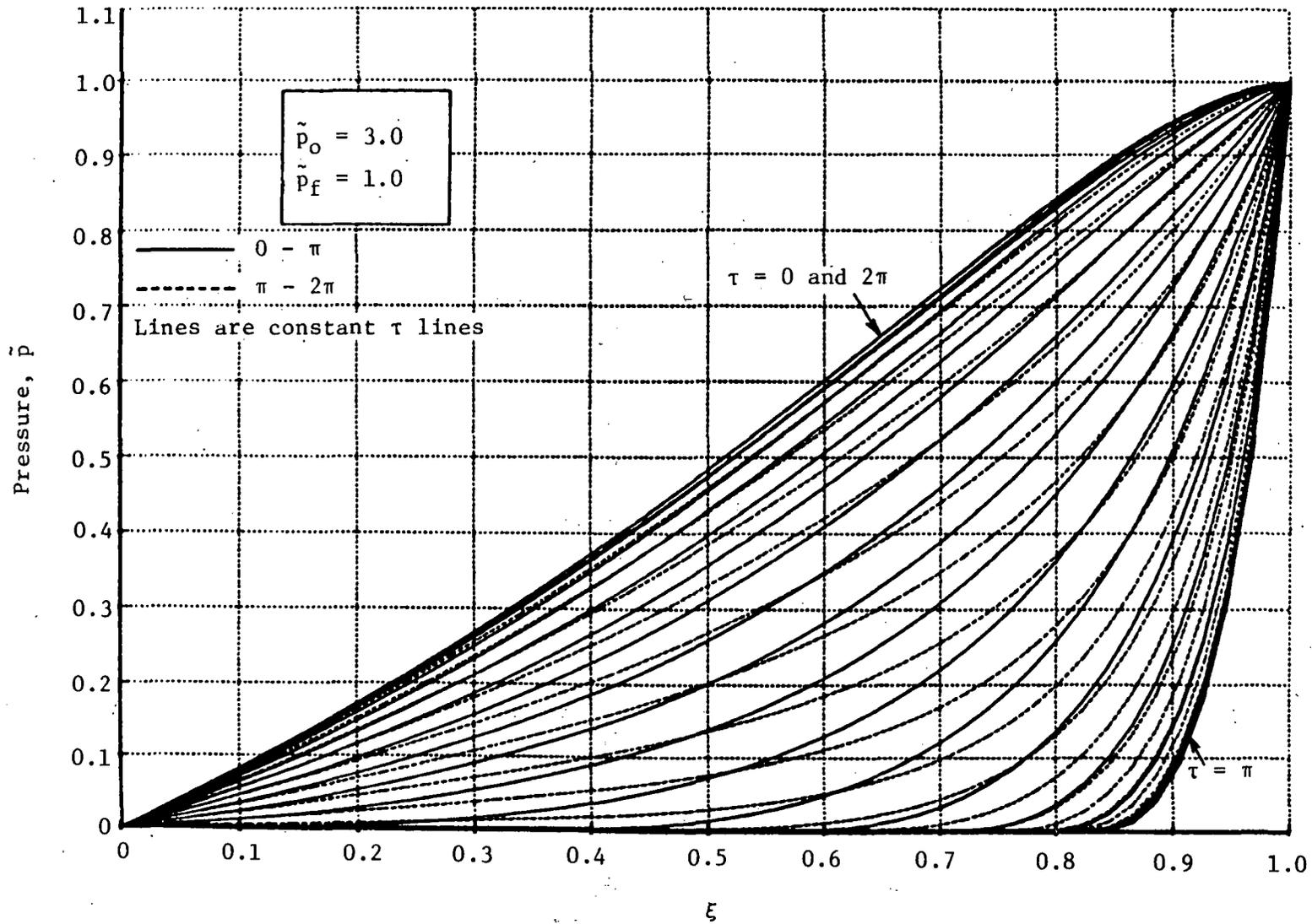
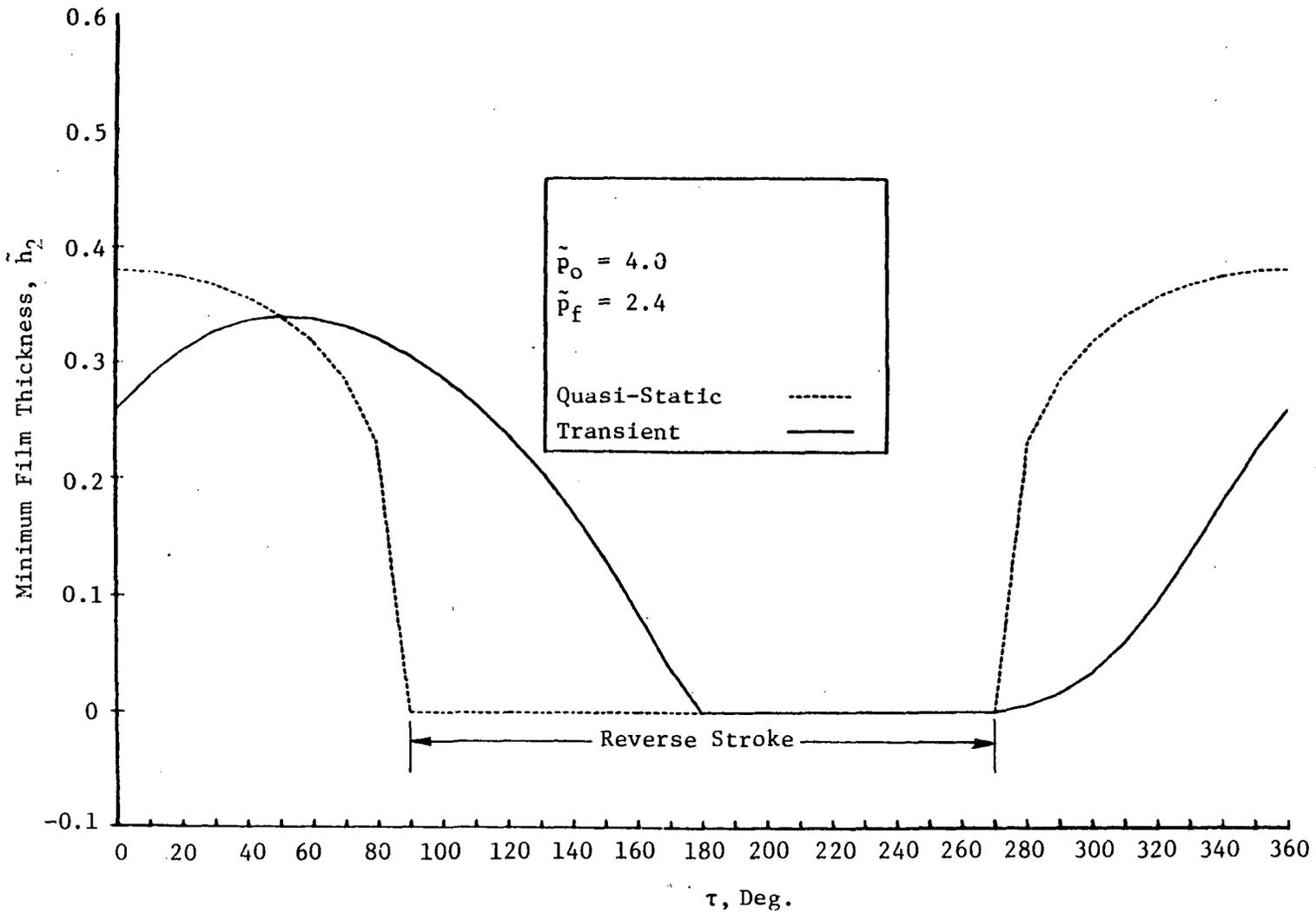


Fig. 3-12 Pressure Profiles as Function of Time for Transient Analysis

Fig. 3-13 Film Thickness h_2 Versus Time

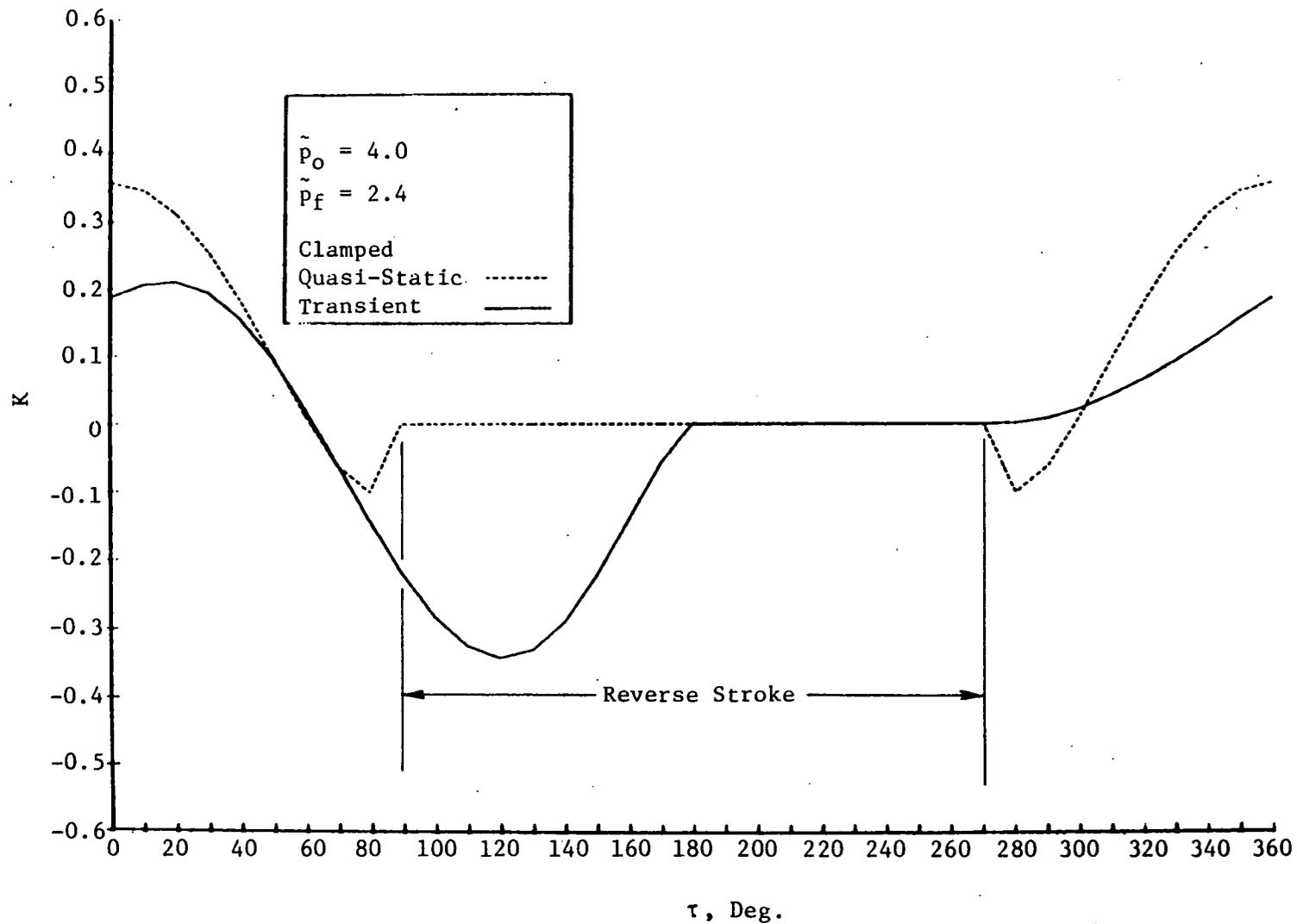


Fig. 3-14 Flow Parameters Versus Time

TABLE 3-1

EFFECTS OF VARIABLE VELOCITY AND SQUEEZE FILM ON FLOW

$\alpha = 0.0282$ $\beta = 0.2569$ $\epsilon = 0.518$ $L_1 = 1.5614$
 $\sigma = 4.0$

\tilde{p}_o	\tilde{p}_f	Analysis*	FORWARD		REVERSE		NET	
			\tilde{Q}	$\Delta\tilde{Q}, \%$	\tilde{Q}	$\Delta\tilde{Q}, \%$	\tilde{Q}	$\Delta\tilde{Q}, \%$
3.0	0	S	0.4542	Ref.	0.1816	Ref.	0.1363	Ref.
		QS	0.4687	+ 3.2	0.1811	0.28	0.1438	+ 5.5
		T	0.4231	- 6.9	0.1731	4.7	0.1250	- 8.3
3.0	1.0	S	0.3392	Ref.	0.4080	Ref.	- 0.0344	Ref.
		QS	0.3524	+ 3.9	0.4059	- 0.8	- 0.0268	22
		T	0.3020	- 11.0	0.4110	+ 0.7	- 0.0545	58
4.0	2.4	S	0.0948	Ref.	0	--	0.0474	Ref.
		QS	0.1348	+ 42	0	--	0.0677	+ 43
		T	0.0711	+ 25	0.1165	--	0.0227	- 148

* S - Static

QS - Quasi-static

T - Transient

- Variable velocity has the effect of increasing the absolute values of all flows; that is, even though it boosts both the forward and back flows, the net flow is increased. Table 3-1 shows this increase to be of the order of several percent for the case of $p_f = 0$. The high percentage changes shown for other cases should not be given too much significance since they center about very low flow levels when a small variation is apt to produce deceptively large effects in terms of percentages.
- The pressures are, of course, positive over the entire ring during the forward stroke, as shown by the $0-\pi$ range of the constant lines on Figure 3-11. Their magnitude, too, follows the velocity curve. During the backstroke, $\pi/2 < \tau < 3\pi/2$, cavitation sets in as high negative velocities are approached ($\tau \rightarrow \pi$); they tend, however, to disappear at the beginning and end of the backstroke, when the negative velocities are low.

The inclusion of squeeze-film effects have the following effect:

- The largest film thickness does not occur at u_{\max} . The peak is delayed so that it is reached after u_{\max} . Likewise, the onset of constant h during the reverse stroke does not commence at $u = 0$, but is also delayed. Thus, squeeze films have the effect of producing a phase shift in the film thickness curve. The shift is fairly large, varying from 50° to 90° in the three cases considered. However, the absolute values of h seem not to be affected.
- The effect on the flow curves is similar in that there is a phase shift with little change in their magnitude. Both the forward and net flows are reduced in comparison to either the steady-state or quasi-static solutions; the amount of back flow tends to increase under the influence of squeeze film forces. Since the quasi-static solution gives higher flows and the static solution gives lower flow than the constant parameter approach, the inclusion of variable velocity without squeeze film effects would yield errors of at least 10%.

- Unlike the behavior in the quasi-static solution, the pressure curves are not identical over the $0-\pi$ and $\pi-2$ halves of the cycle. The pressures are lower when squeeze film effects are included, except over the latter half of the backstroke, when a closing gap helps to increase the pressures above those of the quasi-static analysis.

3.3 Starvation

An estimate of the effect of starvation can be obtained by first considering the clamped case where the area under the bearing land, $0 < x < L$, is predicted to be completely dry at the end of the backstroke. The average film thickness in this area is $\Delta h/2$, and the average Couette flow per unit of circumferential length is $U_0 \Delta h/4$. The time, t_s , for the starved volume to fill up would thus be the cross-sectional area $\Delta hL/2$ divided by the flow per unit of circumferential length. Thus,

$$t_s = (\Delta hL/2)/(U_0 \Delta h/4)$$

The distance traveled prior to flooding is $U_0 t_s = 2L$. This would reduce the effective forward stroke by an amount equal to $2L$. The value of t_s given above will be somewhat low in that it does not account for the development of a resisting pressure gradient that will occur as the starved volume fills up nor does it account for any inertial effects in the entrance region that could impede the start of the filling process. In order to account for these effects when analyzing experimental data, multiply t_s by a factor λ which will in general be greater than or equal to 1. Thus,

$$s_{\text{eff}} = s - 2L\lambda$$

and the dimensionless flow rate K_{eff} becomes

$$K_{\text{eff}} = K (s_{\text{eff}}/s) = K [1 - \lambda(2L/s)] \quad (3-21)$$

The starvation process is, of course, much more complex than that treated here, especially in the unclamped case where there is a partial film and the film is exposed to the sealed pressure, p_0 , so that the cavity could be

trapped in the middle. In order to understand more fully the influence of starvation on pumping ring performance, transparent model experiments should be performed so that the size and behavior of the partial film can be observed and modeled. In the interim, a simple model was adopted to estimate the effect of starvation on pumping for either the clamped or cavitating, nonclamped case.

If the film cavitates during the backstroke at $\xi = \xi_c$, the film thickness at the point, h_c , is

$$h_c = h_2 + \Delta h(1 - \xi_c)$$

and the void volume (volume of gas or vapor in the cavitating region) per unit of circumferential length, V_c , would be

$$V_c = (h_2 + \Delta h - h_c) L \xi_c / 2 = L \xi_c^2 \Delta h / 2$$

If it is assumed that the void volume must fill up before significant pumping can begin and if it fills based on Couette flow at the average film thickness of the cavitating region, the incoming flow per unit of circumferential length would be $U_0 (h_2 + \Delta h + h_c) / 4$, and the starvation time, t_s , would be V_c divided by that flow

$$t_s = (2L\xi_c^2 \Delta h) / [U_0 (h_2 + \Delta h + h_c)]$$

As for the clamped case, the effective forward stroke would be reduced by $\lambda U_0 t_s$. If the dimensionless flow in the forward stroke is denoted by K_f and the backstroke by K_R , then

$$\begin{aligned} K_{f \text{ eff}} &= K_f (s - \lambda U_0 t_s) / s \\ &= K_f \left[1 - 2\lambda \left(\frac{L}{s} \right) \frac{\xi_c^2 \Delta h}{h_2 + \Delta h + h_c} \right] \\ &= K_f \left[1 - 2\lambda \left(\frac{L}{s} \right) \frac{\xi_c^2 \Delta h}{2h_2 + \Delta h (2 - \xi_c)} \right] \end{aligned}$$

and

$$K_{eff} = K_{f\ eff} - K_R \quad (3-22)$$

For the clamped condition, $K_R = 0$, $\xi_c = 1$ and $\tilde{h}_2 = 0$; thus Equation (3-22) reduces to Equation (3-21).

In order to show the effects of starvation, the data for a clamped carbon graphite ring is given in Figure 3-15. The dotted curves include the effects of starvation for a value of $\lambda = 1$. It can be seen that, although the inclusion of the effects of starvation reduces the discrepancy between theory and experiment, it does not resolve the lack of agreement. The effect of values of $\lambda > 1$ will be shown later when comparing theory to experiment.

As a result of the uncertainty regarding the mechanisms of the starvation process, the effects of starvation are not included in the parametric studies in Section 4.0. It is recommended, however, that Equation (3-22) be used to estimate potential effects of starvation when designing pumping rings. The dimensionless starved net flow is given as an output to the computer program RING listed in Appendix B of this report.

3.4 Nonparallel Contours

Essentially, the contours considered are tapered surfaces with a constant slope as shown in Figure 3-16. The reference clearance is that at the trailing edge of the ring; the leading edge clearance is designated by C_M . The slope parameter, δ , is then given by

$$\delta = (dh/dx) = (C - C_M)/L \quad (3-23)$$

In terms of the nondimensional quantities \tilde{h} and ξ , the slope is given by

$$\tilde{\delta} = (d\tilde{h}/d\xi) = (C - C_M)/C = (1 - \tilde{C}_M) \quad (3-24)$$

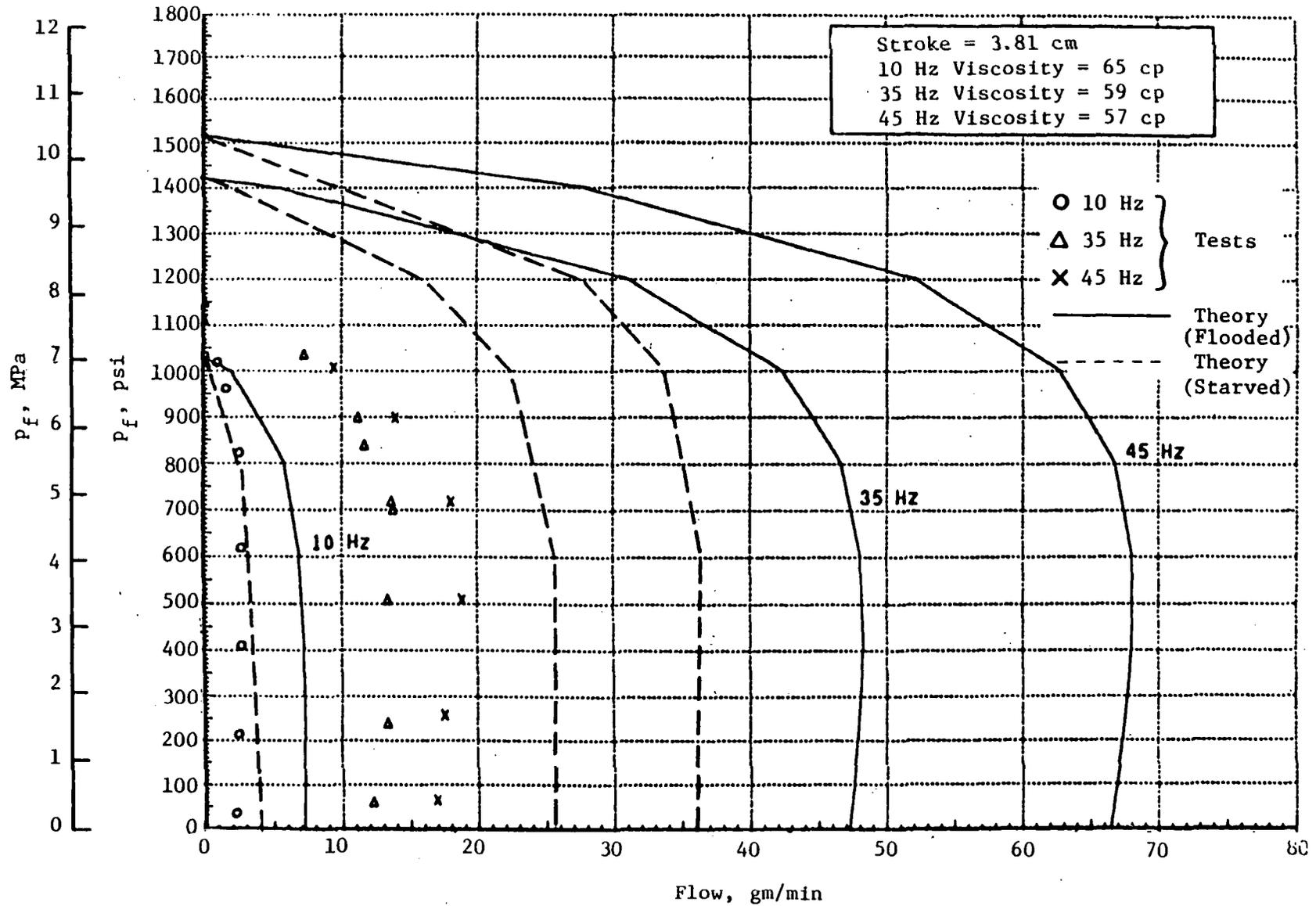
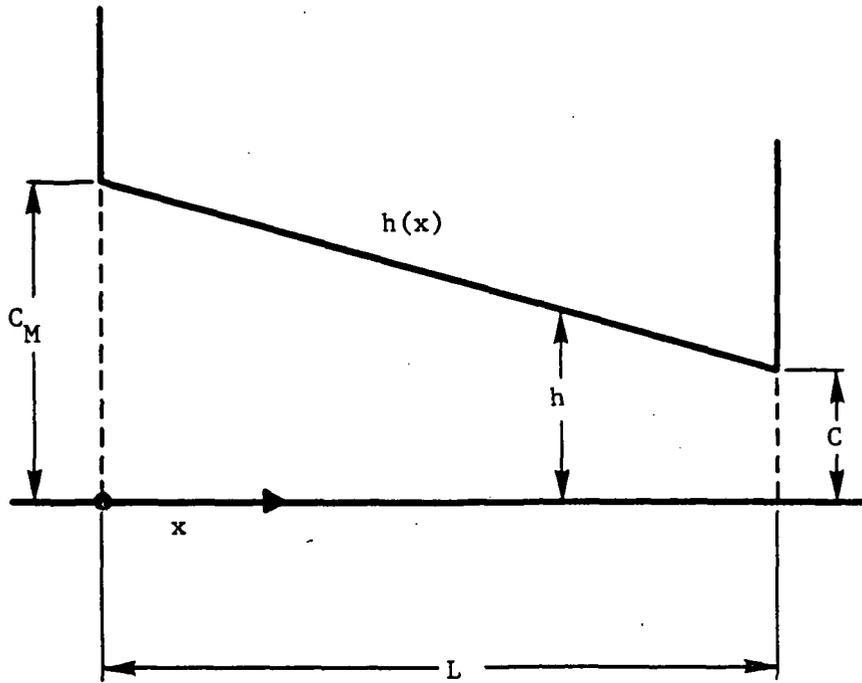


Fig. 3-15 Effect of Starvation on Performance of Carbon Pumping Ring



$$\delta = \frac{C - C_M}{L}$$

$$\tilde{C}_M = (C_M/C)$$

Fig. 3-16 Geometric Taper on Pumping Ring

The clearance ratio $\bar{C}_M = (C_M/C)$ in terms of this δ is then given by:

$$\bar{C}_M = 1 - L\delta/C \quad (3-25)$$

The effect of using tapers on the performance of pumping rings is discussed in Section 5.0. Briefly, it can be said that the use of tapers has little effect, except in cases of low clamping load, p_o , or high values of E , i.e., when there is little elastic deflection of the ring. Clearly, when there is no deflection at all, a taper would constitute the sole mechanism of generating hydrodynamic pressures. In practical applications where deflections are present, the effect of tapers is minimal.

4.0 PARAMETRIC STUDY

A study was made regarding the impact of various structural and operational parameters on the performance of the pumping rings. These parameters include:

- Modulus of Elasticity (E)
- Poisson's Ratio (ν)
- Clearance (C)
- Length (L & L_1)
- Ring Thickness (t)
- Taper (δ)
- Loading (p_0 and e).

The computer runs that provide the results of the parametric study are based on the constant-parameter approach without starvation but including the effects of the backstroke and attendant cavitation. Thermal and transient effects were omitted because, as shown on Figure 4-1, their effect is not of the sort as to qualitatively change the behavior of the pumping ring. The constant parameter analysis should be sufficient to reveal the essential features of the parametric relationships without the undue complications of the more elaborate analysis.

Since it would ultimately be desirable to optimize the design of pumping rings, the question arises as to what constitutes an optimum. Given the purposes for which these devices are customarily used, an optimum ring was deemed to be one which, within given constraints such as reasonable values of p_0 , C, etc., can maintain the highest possible sealing pressure without excessive wear. This implies the highest values of p_f and a low, preferably zero, frictional force F. Thus, a ring design which generates high p_f and just clamps shut upon reversing the stroke would fulfill this criterion. An additional quantity of interest, perhaps, is also the maximum flow Q_0 (flow at $p_f = 0$), that such a ring can produce. Thus, in the plots that follow, the items of p_f , F, and Q_0 will be the items against which ring performance will be measured.

The parametric study was conducted by first establishing a standard or reference design and then by varying its parameters, one at a time, to values below

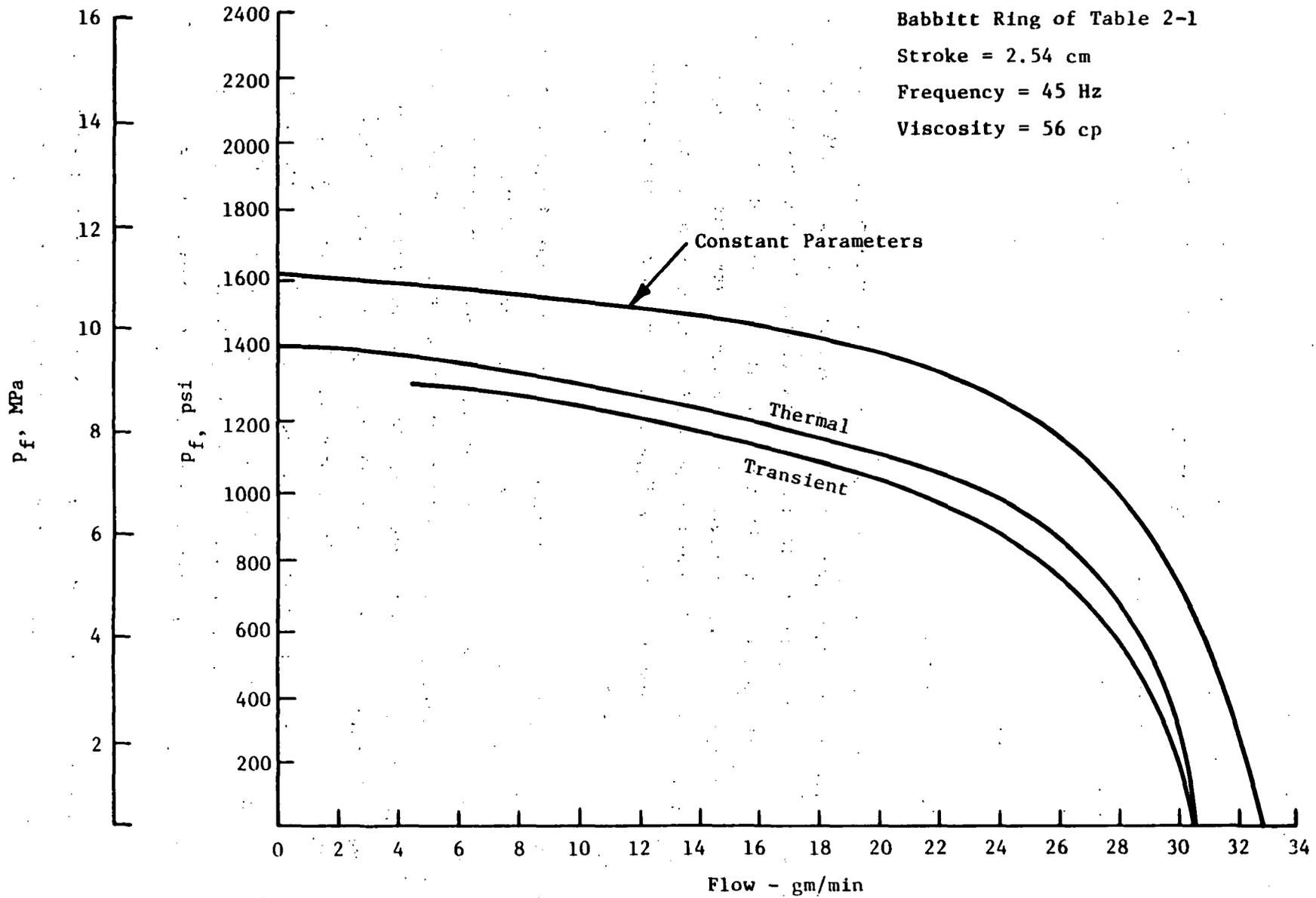


Fig. 4-1 Modifications Due to Thermal and Transient Effects

and above the reference quantities. The parameters for this standard ring are given in Table 4-1 along with selected departures from the standard set. Including the standard, most parameters have four computed points from which optimization plots can be made. The results of the parametric study are summarized in Table 4-2; more relevant plots are portrayed in Figures 4-2 through 4-11. In Figure 4-4, the reason the film thickness h_2 drops with an increase in clearance, is due to the drop in p_{fm} ; a high p_{fm} usually helps to contract the clamping pressure and thus maintain a high h_2 . In Figure 4-6, the explanation for an increase in flow with a rise in length L is that with a greater L the hydrodynamic effects are increased, and thus, also the flow. As seen, four parameters, the lengths L and L_1 , taper δ , and Poisson's ratio have little effect on performance. Of the two important variables, namely, clearance and clamping force, the first has to be raised above its optimum for maximum p_f , and the latter reduced below the optimum, if large frictional forces are to be avoided. Since the clamping force is given by the product of p_o and e , Figure 4-12 shows the effects of using the same clamping force, 31.96 KN/m (182.5 lb per in.) of circumference, but by varying the relative values of p_o and e . As seen, higher values of p_f are achieved when p_o is high, while higher values of Q_o are achieved when e is high. The differences, however, are not striking.

It should be pointed out that the particular optima recorded in Table 4-2, such as $t = 2.54$ mm (0.1 in.), $C = 25.4$ microns (1 mil), etc., are valid in the range of parameters characterizing the particular ring specified in Table 4-1. Thus, for example, for a material with a much higher E value, optimum results would be achieved with lower values of clearance and ring thickness. The opposite would be true for a material with a value of E much below the 34.5 GPa (5×10^6 psi) assigned to the standard design.

TABLE 4-1

STANDARD DESIGN FOR PARAMETRIC STUDY

$p_o = 10.3 \text{ MPa (1500 psi)}$
 $e = 3.175 \text{ mm (0.125 in.)}$
 $T = 49^\circ\text{C (120}^\circ\text{F)}, \mu = 59 \times 10^{-3} \text{ Pa-sec, (8.6} \times 10^{-6} \text{ Reyns)}$
 $E = 3.45 \cdot 10^4 \text{ MPa (5} \cdot 10^6 \text{ psi)}$
 $\nu = 0.36$
 $s = 12.7 \text{ mm (0.5 in.)}$
 $f = 35 \text{ Hz}$
 $R = 12.7 \text{ mm (0.5 in.)}$
 $L = 6.35 \text{ mm (0.25 in.)}$
 $L_1 = 10.2 \text{ mm (0.4 in.)}$
 $C = 0.019 \text{ mm (0.75} \times 10^{-3} \text{ in.)}$
 $t = 1.9 \text{ mm (0.075 in.)}$
 $\delta = 0$

VARIATIONS IN PARAMETERS

C: 0.0063 mm, 0.0127 mm, 0.0381 mm (0.25×10^{-3} in., 0.5×10^{-3} in., 1.5×10^{-3} in.)
E: $6.895 \cdot 10^3$ MPa, $20.7 \cdot 10^3$ MPa, $68.95 \cdot 10^3$ MPa, $207 \cdot 10^3$ MPa (10^6 psi, $3 \cdot 10^6$ psi, $10 \cdot 10^6$ psi, $30 \cdot 10^6$ psi)
L: 5.1 mm, 7.6 mm, 10.2 mm (0.2 in., 0.3 in., 0.4 in.)
 p_o : 5.17 MPa, 6.895 MPa, 13.79 MPa (750 psi, 1,000 psi, 2,000 psi)
 ν : 0.25, 0.5
t: 1.27 mm, 2.52 mm, 3.81 mm (0.05 in., 0.1 in., 0.15 in.)
L: 6.35 mm, 15.2 mm (0.25 in., 0.6 in.)
 δ : -10^{-3} , $-2 \cdot 10^{-3}$, $-3 \cdot 10^{-3}$

TABLE 4-2

OPTIMUM PARAMETERS FOR STANDARD RING

(See Table 4-1)

ITEM		RANGE EVALUATED	For P _{fm}	FOR Q _o
			SUBJECT TO F=0	
E	psi 10 ⁻⁶	1 - 30	5.5	6
	MPa 10 ⁻³	6.89 - 207	41.3	41.3
v		0.25 - 0.5	No Effect	No Effect
C	mils	0.25 - 1.5	1.0	1.0
	microns	6.35 - 38.1	25.4	25.4
L	in	0.2 - 0.4	No Effect	Highest
	mm	5.08 - 10.16		
L ₁	in	0.25 - 0.6	No Effect	No Effect
	mm	6.35 - 15.2		
t	in	0.05 - 0.15	0.100	0.102
	mm	1.27 - 3.81	2.54	2.64
δ		0 - (-3·10 ⁻³)	No Effect	0
p _o *	psi	750 - 2,000	1100	1050
	MPa	5.17 - 13.8	7.5	7.20

*See Also Fig. 4-12.

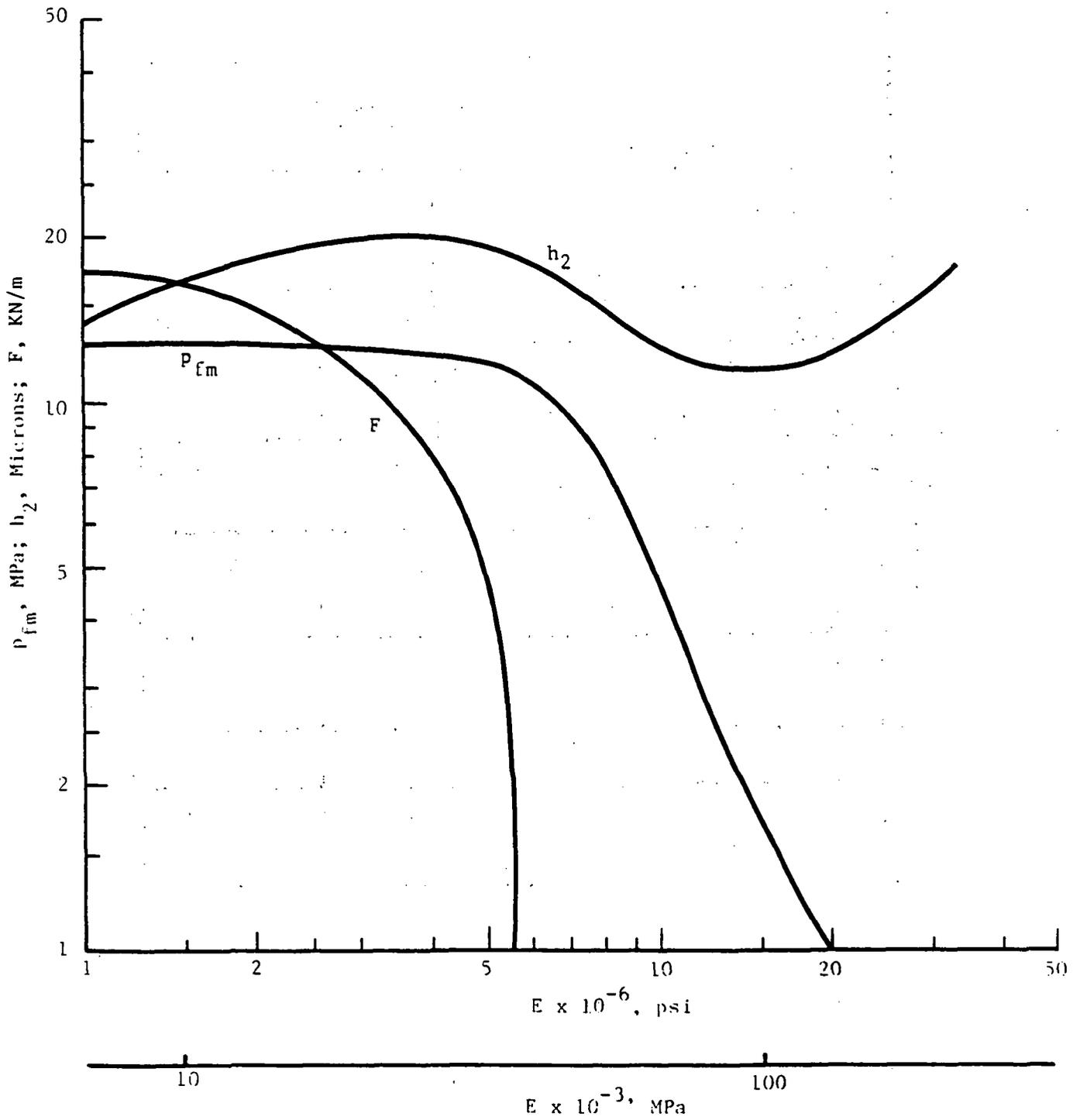


Fig. 4-2 Effect of E on Performance Standard Ring (See Table 3-1)

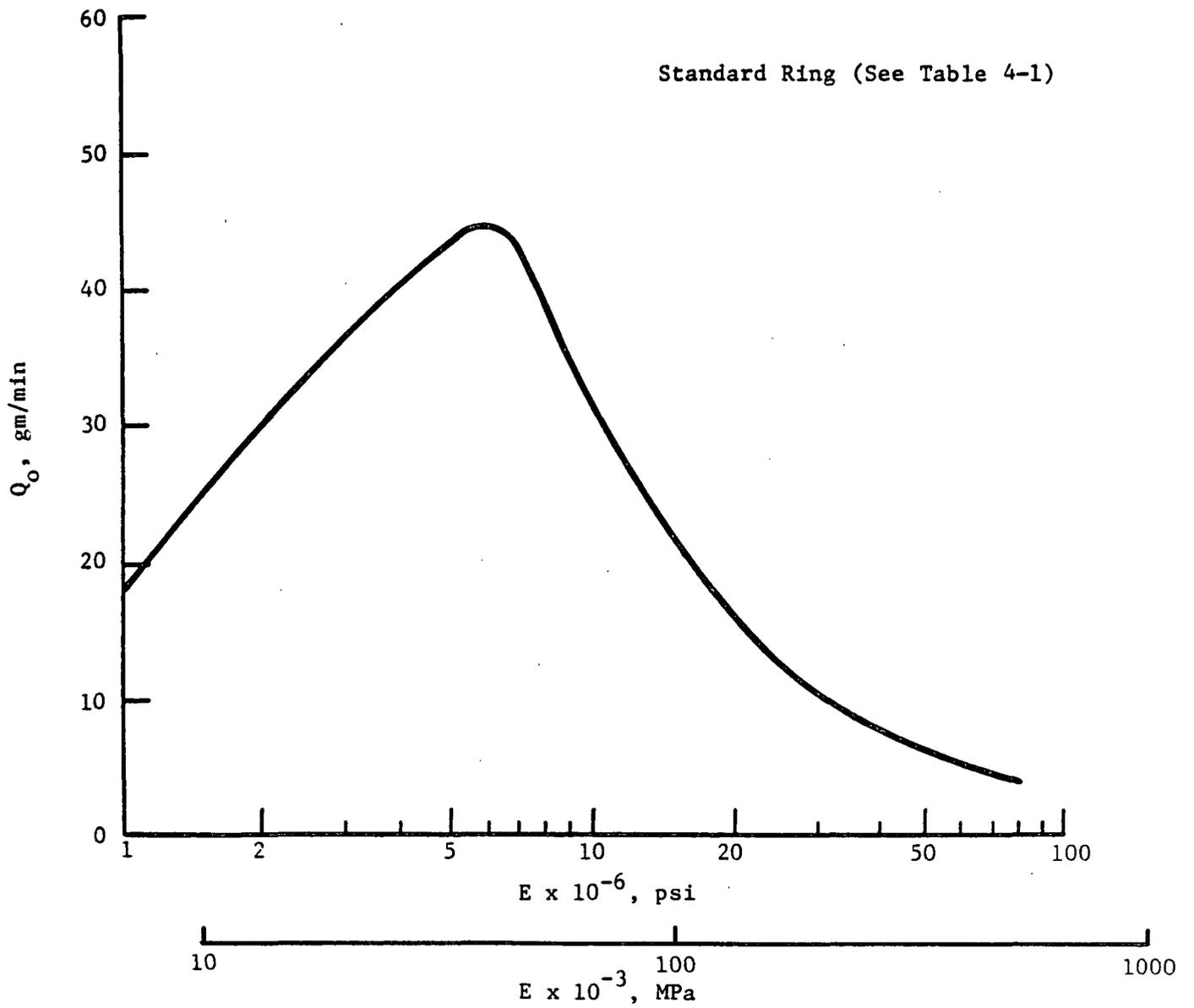


Fig. 4-3 Effect of E on Q_o

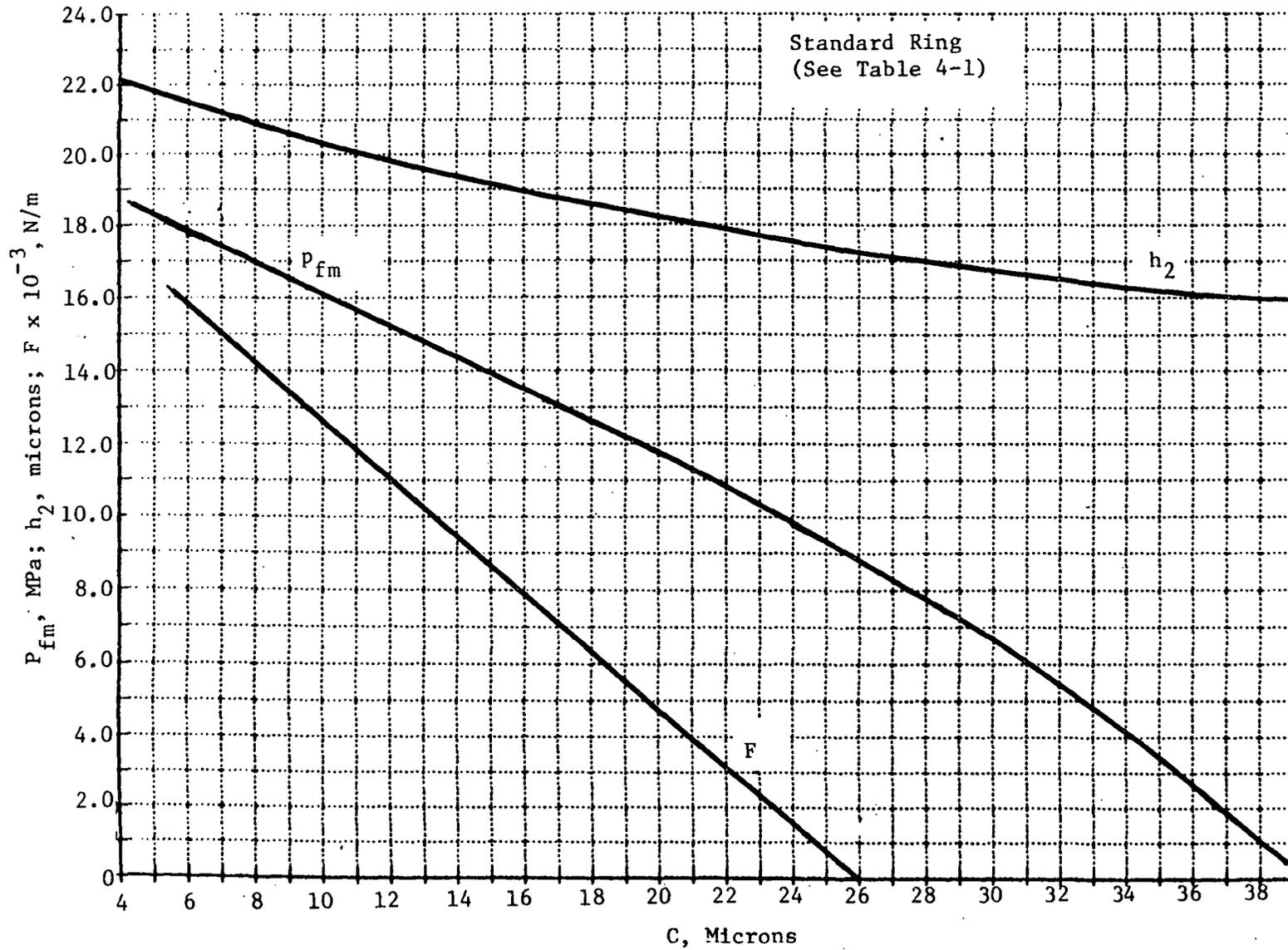


Fig. 4-4 Effect of C on Performance

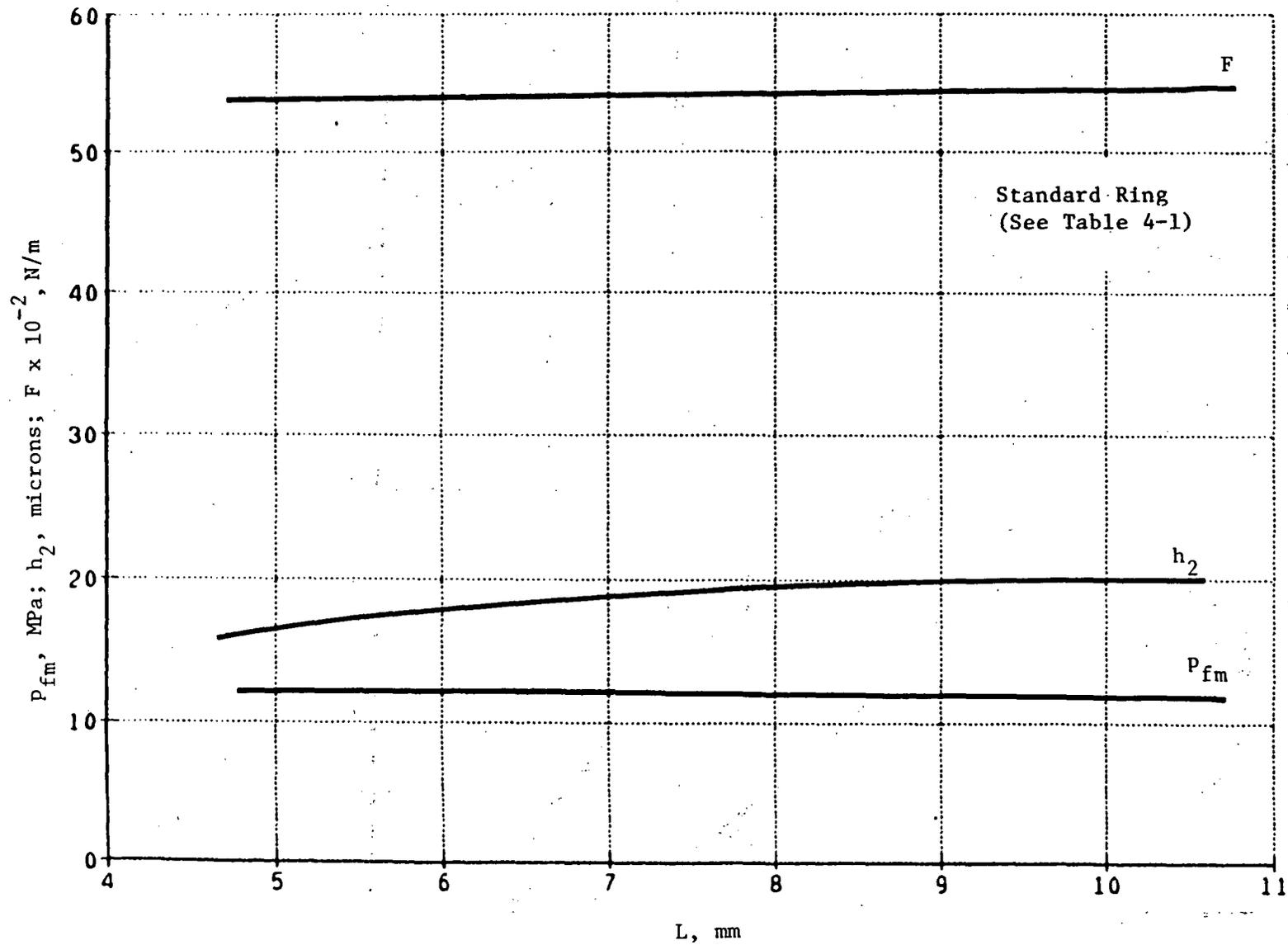


Fig. 4-5 Effect of L on Performance

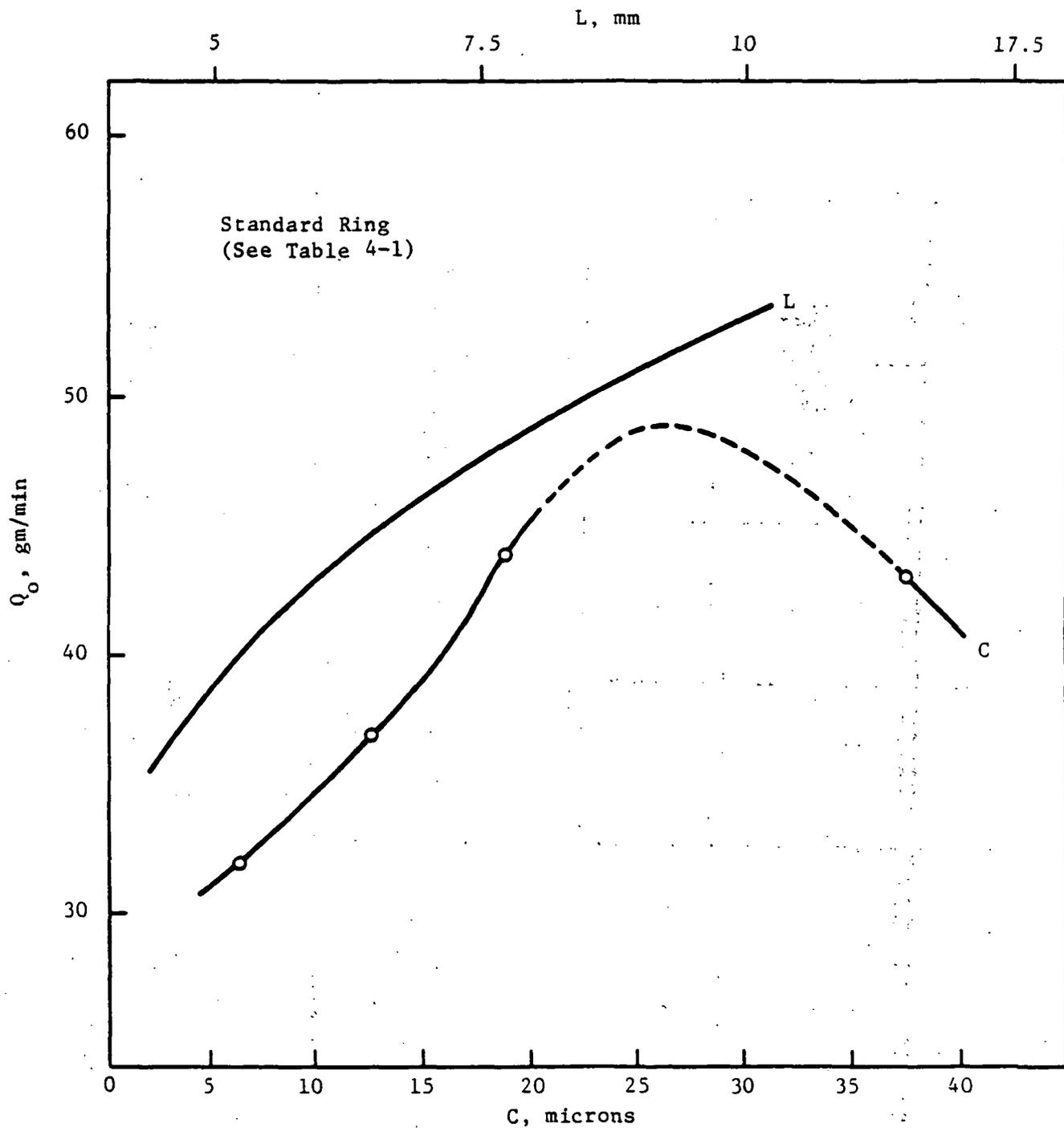


Fig. 4-6 Effects of C and L on Q_o

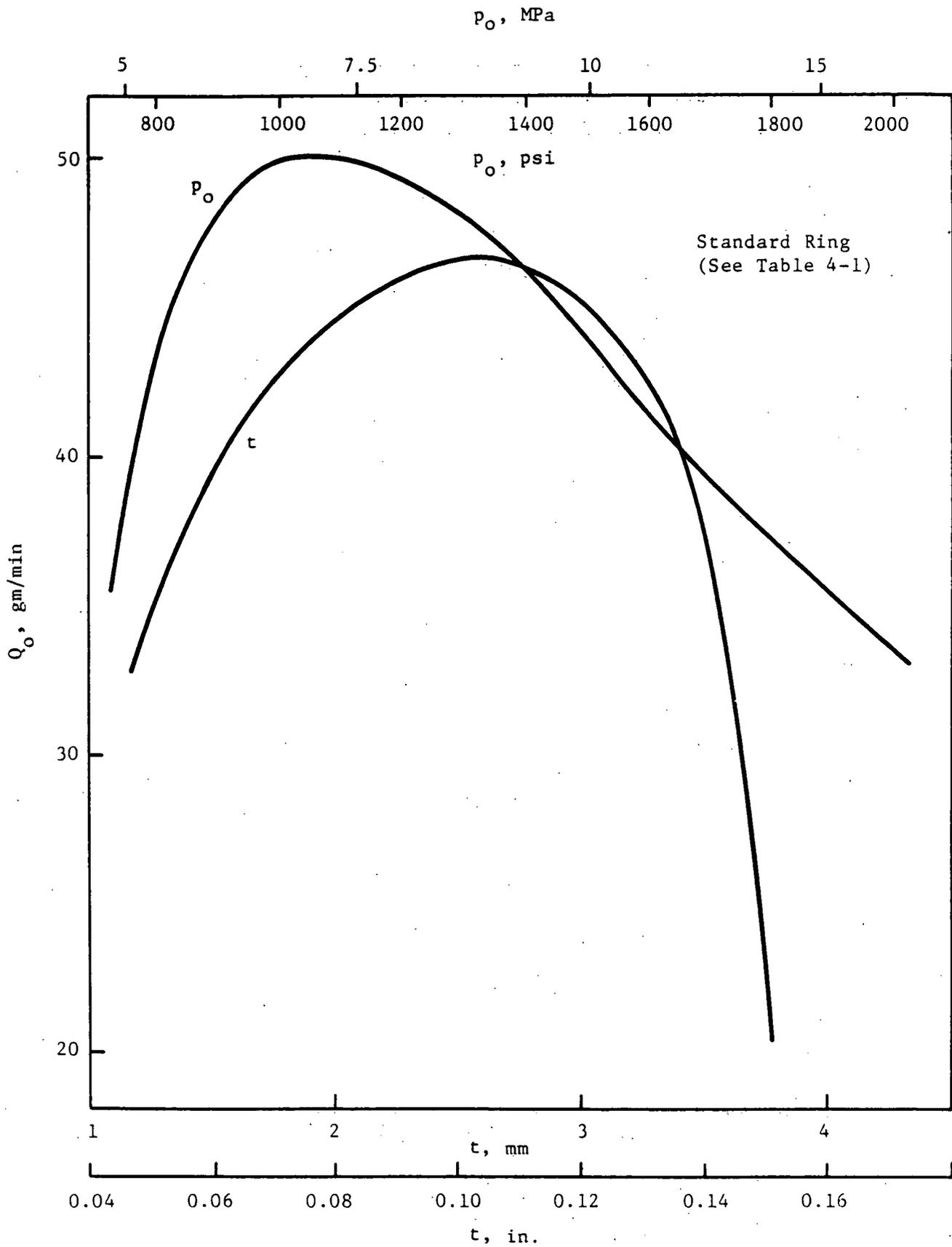


Fig. 4-7 Effects of t and p_o on Q_o

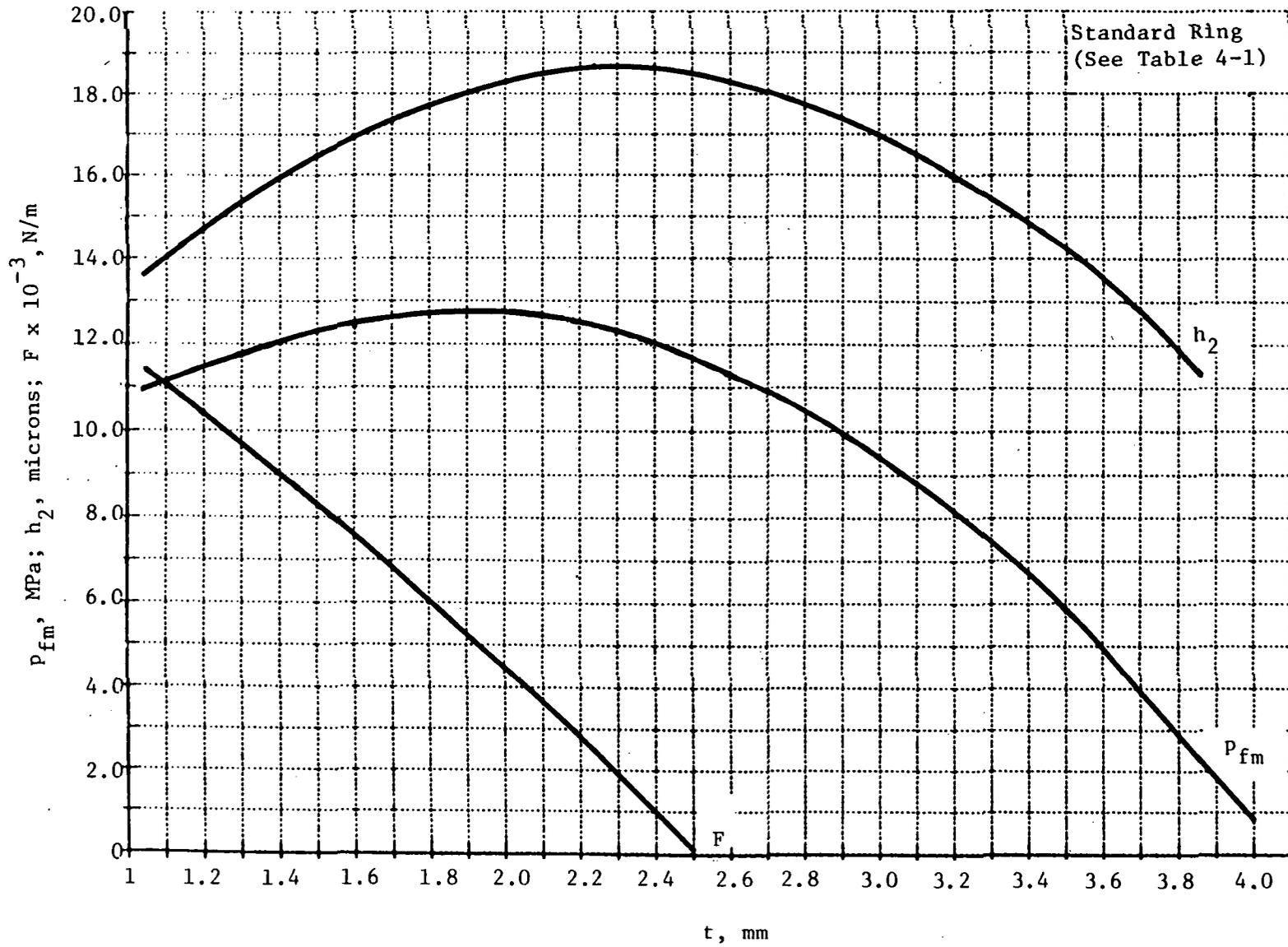


Fig. 4-8 Effect of t on Performance

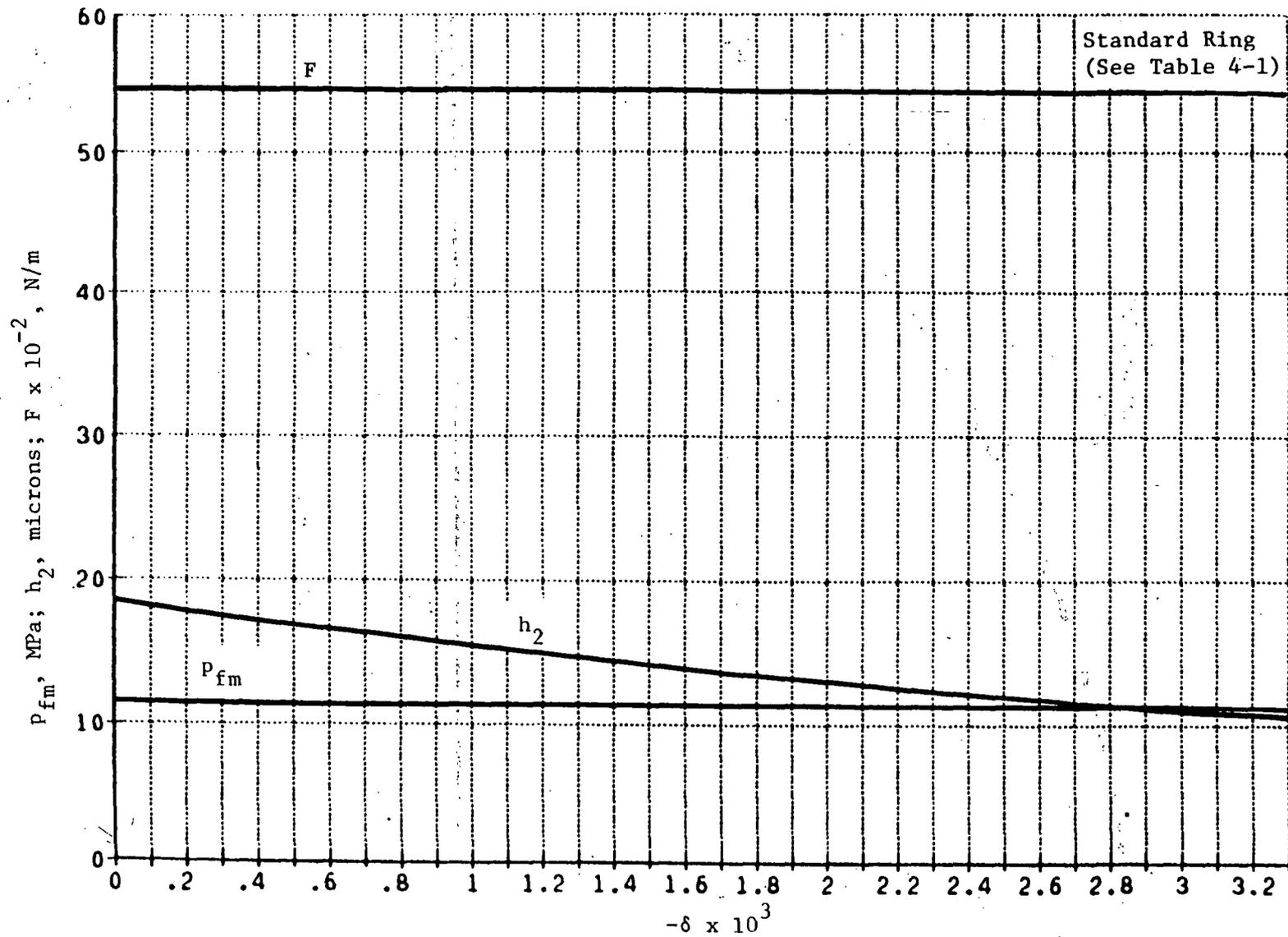


Fig. 4-9 Effects of Taper on Performance

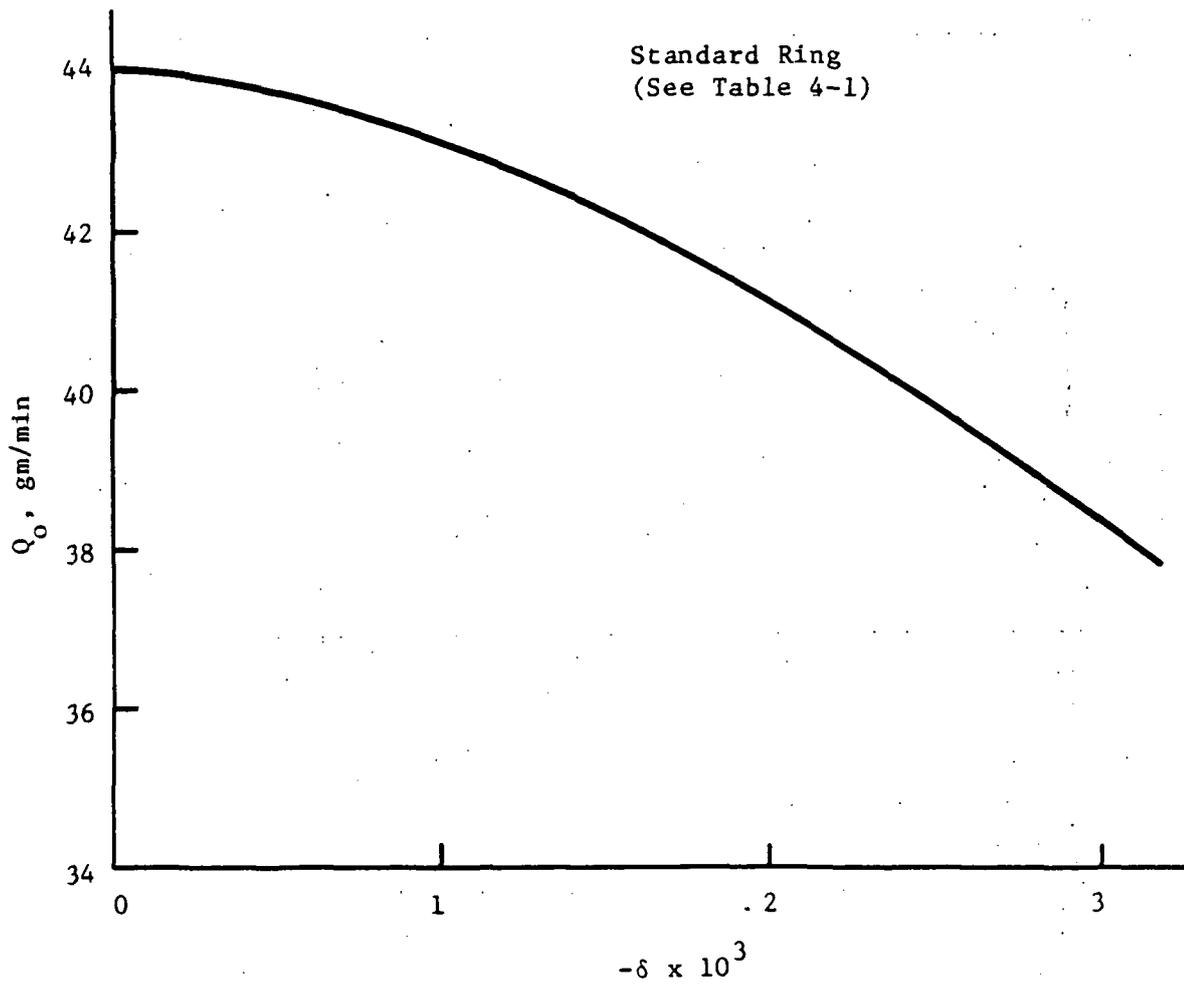


Fig. 4-10 Effect of δ on Q_0

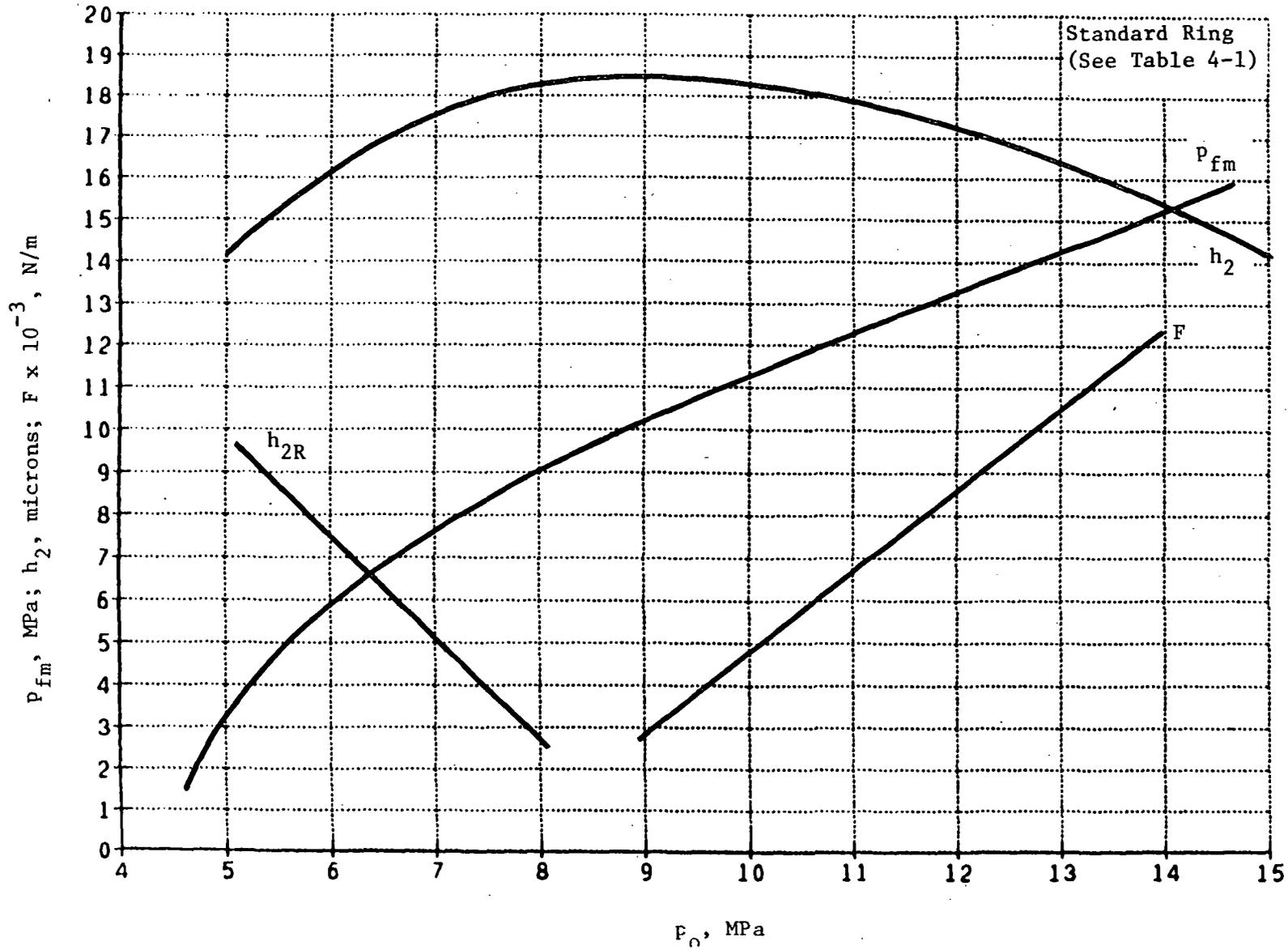


Fig. 4-11 Effect of p_o on Performance

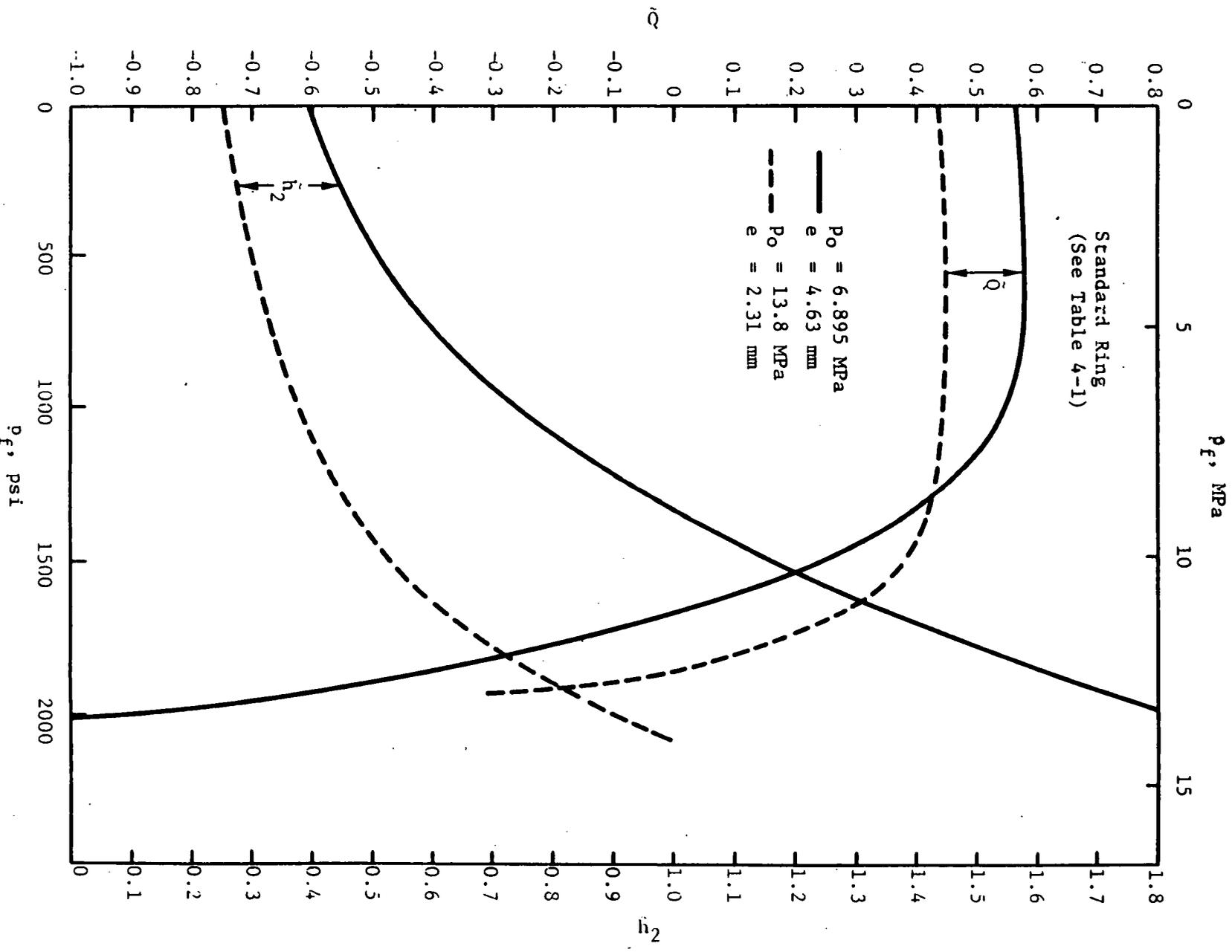


Fig. 4-12 Combined Effect of p_o and e

5.0 EXPERIMENTAL PROGRAM

Following the optimization of the candidate rings, tests were run on three rings to verify their performance and their agreement with theoretical predictions. The test rig and the experimental procedures were the same as those used on the tests described in Ref. [1].

The pumping ring program, given in Appendix B based on the analysis performed here, has been used to select ring designs for experimental testing. Initially, a set of studies was performed with steel, babbitt, and Rulon J rings as examples of a high, medium, and low modulus pumping ring. Overall dimensions were selected consistent with the experimental test rig. The principal design variables to be evaluated within the applied pressure, p_o , the clearance, c , and the average thickness, t . A target value of p_o of 1500 psi was used; however, it was found not feasible to design to this pressure with the low modulus Rulon ring. Sample geometries that have been arrived at are shown in Table 5-1. The general configuration of the pumping rings submitted for test is shown in Figure 5-1.

The principal criterion used involves being able to significantly deflect the ring to obtain pumping action without excessive clamping during the backstroke, which would result in excessive wear.

Based on the criterion above, the design of the steel ring would require relatively small thickness and a small clearance to obtain suitable compliance and deflection characteristics. The tolerances required seem excessive for the present application, and it is thought that steel rings would only be applicable in situations requiring much higher pressures. It was thus decided to confine testing to medium and low modulus materials ranging between babbitt, carbon graphite, and Rulon. The test rig and experimental procedures were the same as those used on the test.

One problem area revealed in the course of testing concerns the clamping load. Difficulties were encountered in testing the carbon and Rulon rings, and often such attempts at testing led to breakage of the specimens. It was also noted that some of the babbitt rings were severely worn, not at the downstream end

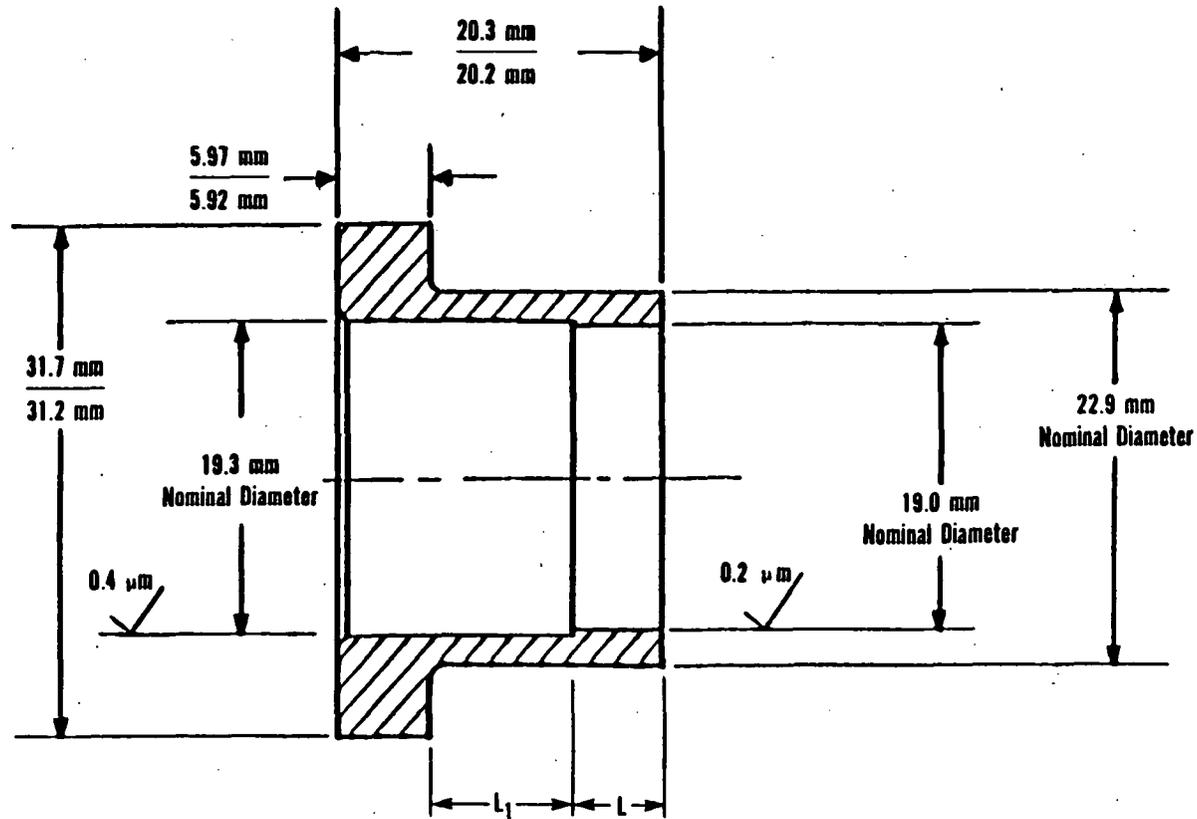


Fig. 5-1 Hydrodynamic Oil Pumping Ring Dimensions

TABLE 5-1

SAMPLE GEOMETRIES AND CONDITIONS FOR
PARAMETRIC STUDY OF CANDIDATE TEST RINGS

$L = 6.73 \text{ mm}$ (0.265 in.), $L_1 = 7.62 \text{ mm}$ (0.30 in.), $\delta = 0$, $s = 38.1 \text{ mm}$ (1.5 in.),

$f = 35 \text{ Hz}$.

ITEM	STEEL	BABBITT	RULON J
R, mm (in.)	9.5 (0.38)	9.5 (0.38)	9.5 (0.38)
E, MPa (psi)	$207 \cdot 10^3$ (30×10^6)	$51 \cdot 10^3$ (7.5×10^6)	$1.72 \cdot 10^3$ (0.25×10^6)
ν	0.3	0.36	0.46
p_0 , MPa (psi)	10.34 (1500)	10.34 (1500)	5.17 (750)
C, microns (mils)	8.96 (0.35)	19 (0.75)	50.8 (2)
t, mm (in.)	1.27 (0.05)	1.90 (0.075)	3.81 (0.15)

where wear might be anticipated but far upstream, ahead of the actual pumping area, as shown in Figure 5-2a. This led to the conclusion shown in Figure 5-2b that the high-pressure oil used for clamping leaked past the O-ring and, by pressing against bracket A, effectively produced a sealed chamber over the OD of the ring. The pumping ring was thus loaded with a pressure, p_o , over the entire length ($L_1 + L$) instead of only the narrow region, e , covered by the O-ring. This unprecedented pressure loading gave the ring the flared shape shown in Figure 5-2b which led to high wear upstream of the active pumping area and to breakage of the weaker rings.

The problem was resolved by opening relief passages behind bracket A so that any fluid leaking past the O-ring would be scavenged outside. Since the leakage past the O-ring occurs only when p_f approaches p_o , such leakage should not affect the build-up of sealed pressure p_f at levels below p_o . With the relief grooves in place, all unsatisfactory tests were repeated; no further difficulties in testing the Rulon and carbon rings were encountered. It should also be noted here that the sealed pressure p_f was not observed to significantly exceed the loading pressure p_o . This is believed to be a result of leakage past the O-ring, which occurs when p_s starts to exceed p_o , as indicated in Figure 5-3.

The rings tested were made of either babbitt, carbon graphite, or Rulon. The geometry and dimensions of the various rings tested are as summarized in Table 5-2 and, except for one item (Item VII), had the following two dimensions in common:

Shaft Diameter - 19.05 mm

Inside Diameter of Back Section (over length L_1) - 19.3 mm

The viscosities of the oil used are those given in Figure C-1 of Appendix C.

5.1 Tests with Large Babbitt Ring

The 19.05-mm diameter babbitt ring was tested at three frequencies, 60, 35, and 10 Hz, and at two strokes, 50.8 mm and 25.4 mm. The parameters investigated were the pumping ring length (L), clearance (C), and the effect of a taper (δ). Tables C-1 through C-4 in Appendix C give the detailed results of

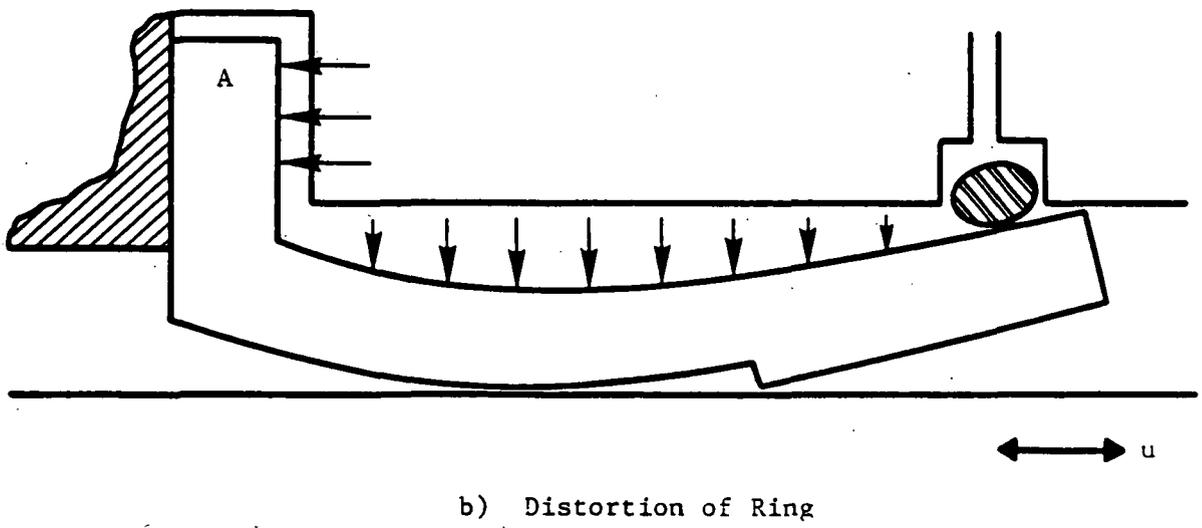
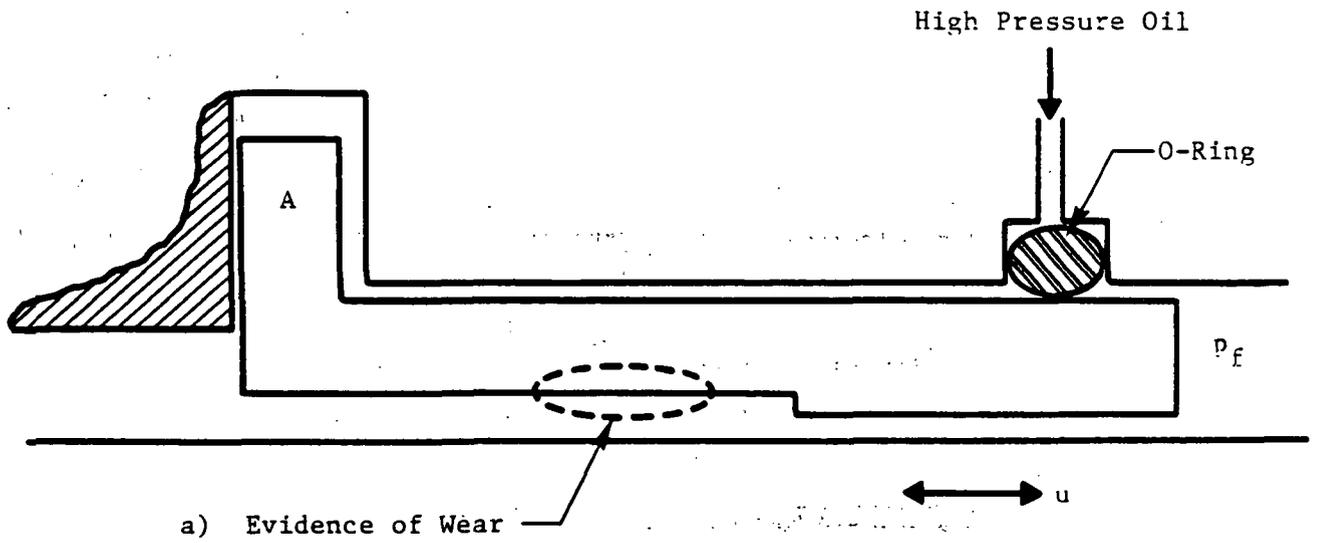


Fig. 5-2 Leakage of High Pressure Oil Past O-Ring

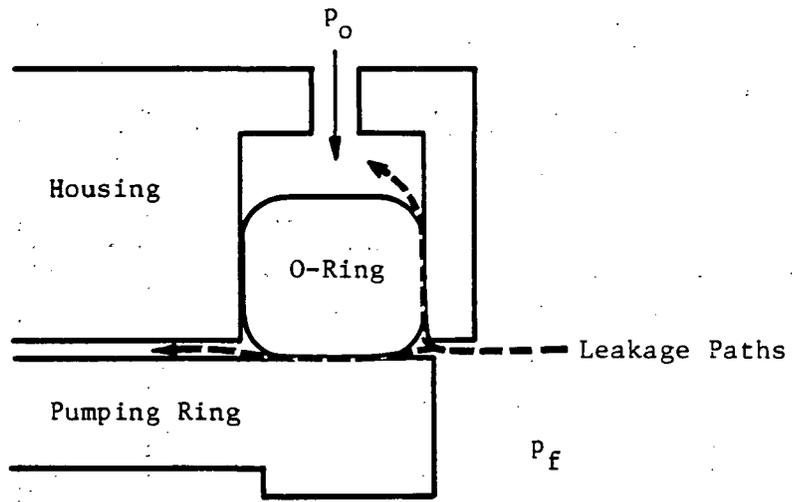


Fig. 5-3 Schematic of Leakage Past O-Ring at High Sealed Pressures

TABLE 5-2

LIST OF PUMPING RINGS TESTED

I.D. NO.	MATERIAL	R mm (in.)	O.D. mm (in.)	L ₁ mm (in.)	L mm (in.)	C microns (mils)	- $\delta \times 10^3$	TEST I.D. NO.
I	Babbitt	9.52 (.375)	21.6 (.850)	7.56 (0.298)	6.78 (.267)	11.43 (0.45)	0	A-1-A-1
II	↓	↓	21.5 (.846)	↓	6.78 (.267)	19.05 (0.75)	↓	A-2-A-1
III	↓	↓	21.4 (.843)	↓	4.83 (.190)	12.70 (0.50)*	↓	A-1-B-1
IV	↓	↓	21.4 (.843)	↓	6.78 (.267)	25.4 (1.0)*	3.75	A-1-A-2
V	Carbon	9.52 (.375)	22.07 (.869)	7.56 (0.298)	6.78 (.267)	20.32 (0.8)	0	C-1-A-1
VI	Rulon	9.52 (.375)	24.15 (.951)	7.56 (0.298)	6.78 (.267)	42.55 (1,675)*	↓	B-1-A-1
VII	Babbitt	6.00 (.236)	13.92 (.548)	5.59 (0.22)	5.08 (.200)	8.89 (0.35)	↓	D-1-A-1

* Average of two rings.

the tests. The self-pressurization runs represent an arrangement in which the clamping pressure was provided by the sealed pressures generated by the pumping ring. This represents the case of $p_o = p_f$ at any instance. It turned out that, in order to start the untapered pumping ring, a priming pressure was required (that is, when at the start, $p_f = 0$, no self-clamping was possible). Some initial $p_{fo} > 0$ had to be provided by external means in order to enable the pumping ring to apply self-pressurization and build up a p_f beyond the initially supplied priming pressure, p_{fo} . The tapered rings were found to be self-priming.

With regard to the effect of the several variables tested, the plots suggest the following:

- Effect of Clearance - Table 5-3 shows the changes in maximum flow and pressure induced by changing the clearance from 11.4 microns to 19 microns. These changes are minimal, indicating that the optimum lies within the range of values.
- Effect of Length - As shown in Table 5-4, there is a definite advantage to a higher length, L , with regard to maximum flow rates; but the shorter rings produced higher maximum levels of sealed pressure, p_{fm} .
- Effect of Taper - Table 5-5 confirms what can be intuitively deduced as the desirability of having a geometric taper. At high clamping pressures when ring deflections are large, the geometric taper is counter-productive. However, at $p_o = 3.45$ MPa, a taper produces higher flow rates and higher levels of maximum pressure.

5.2 Tests with the Rulon and Carbon Ring

The detailed results of the tests on the Rulon and carbon rings are given in Tables C-5 and C-6 in Appendix C. In both rings, the higher clamping pressures produced higher reservoir pressures, p_{fm} , but the lower values of p_o yielded higher maximum flow rates. This trend was already noticeable with the babbitt ring, but as the value of E drops, it becomes more pronounced so that,

TABLE 5-3

EFFECT OF CLEARANCE

RUNS I AND II

P _o MPa (psi)	s mm (in.)	f, Hz	Q _o , gm/min		P _{fm} , MPa	
			C = 0.0114 mm (0.45 mil)	C = 0.019 mm (0.75 mil)	C = 0.0114 mm (0.45 mil)	C = 0.019 mm (0.75 mil)
8.62 (1250)	50.8 (2)	60	43	42	9	9
		35	21	22	9	9
		10	4	4	8.8	8.8
6.895 (1000)	50.8 (2)	60	50	51	7.2	7.2
		35	24	27	7.2	7.2
		10	4	4	7.2	7.2
8.62 (1250)	25.4 (1)	60	7	7.5	8.8	9.2

TABLE 5-4

EFFECT OF REDUCED LENGTH L

RUNS I AND III

P _o MPa (psi)	s mm (in.)	f. Hz	Q _o , gm/min		P _{fm} , MPa	
			L=6.78 mm (0.267 in.)	L=4.83 mm (0.19 in.)	L=6.78 mm (0.267 in.)	L=4.83 mm (0.19 in.)
6.895 (1000)	50.8 (2)	60	50	34	7.2	7.8
		35	25	17	7.2	7.8
		10	4	2.5	7.2	7.0
3.45 (500)	50.8 (2)	60	59	51	3.8	7.2
		35	28	27	3.8	7.2
		10	3.5	3.5	3.8	7.2
8.62 (1250)	25.4 (1)	60	70	60	8.8	8.0

TABLE 5-5
EFFECT OF TAPER
RUNS I AND IV

P _o MPa (psi)	s mm (in.)	f, Hz	Q _o , gm/min		P _{fm} , MPa	
			δ = 0	δ = -3.75 · 10 ⁻³	δ = 0	δ = -3.75 · 10 ⁻³
8.62 (1250)	50.8 (2)	60	42	32		
		35	22	17	9.2	9.2
		10	4	4	8.9	9.0
6.895 (1000)	50.8 (2)	60	50	42	9.2	7.2
		35	24	18		
		10	4	2		7.5
3.45 (500)	50.8 (2)	60	60	> 80	3.8	4.2
		35	28	38	3.8	4.2
		10	4	4	3.8	3.8
8.62 (1250)	25.4 (1)	60	7	2.5	8.8	8.8

for the Rulon rings, Q_o at $p_o = 1.72$ MPa (250 psi) is two or three times the value of Q_o at $p_o = 5.17$ (750 psi).

5.3 Tests with the Small Babbitt Ring

The results for the 6-mm radius babbitt ring are given in Table C-7. The diameter was reduced roughly 1/3 as compared to the large rings. Neither the flows, Q_o , nor the maximum sealed pressure, p_{fm} , were much affected by the change in size.

6.0 DISCUSSION OF RESULTS

Three sets of comparative results are presented in Appendix D.

- The babbitt rings, which comprise the large size with two different clearances and a small size ring
- The carbon graphite ring
- The Rulon ring.

The theoretical curves presented in Appendix D were obtained without the starvation correction and without allowance for the fact that the sealed pressure does not exceed the loading pressure due to leakage past the O-ring.

A number of qualitative conclusions can be drawn from a study of the figures in Appendix D. The major discrepancy observable is the fact that the measured flows are substantially smaller than those predicted by theory at zero sealed pressure. This is believed to be partially due to effects of starvation as will be discussed later. For practical purposes, this discrepancy is not believed to be a major one in that the pumping rings will normally be used to develop a buffer pressure as opposed to being designed to deliver prescribed flow rate. The predicted maximum developable pressure at zero net flow was found to be in better agreement with the experimental data. Again, all of the curves that predict sealed pressure higher than the loading pressures should be truncated at the loading pressure value due to leakage past the secondary seal.

It is important to note that all of the data presented in Appendix D was obtained with seals that were sized and designed with the analysis contained here. In all cases, as predicted, the seals did perform the task of developing and maintaining the designed sealed pressures. Significant pressures could not be developed for steel rings where the theory indicated that sufficient deflection would not be obtainable to provide adequate pumping. Although the analysis presented so far does not accurately predict deliverable flow, it can be used reliably to size and design pumping rings. In order to obtain and improve fit to the flow data, a two-constant empirical relationship has been determined from the data for babbitt pumping rings to attempt to

account for starvation. The results of this fit when compared with experimental data are described below.

6.1 Empirical Correction for Starvation

An empirical relationship for the starvation factor, λ , as a function of L/s of the form

$$\lambda = 0.74 (s/L)^{0.58}$$

has been obtained by fitting flow rates at zero sealed pressure for ring No. I at $p_0 = 8.62$ MPa. Comparisons have been made for rings I - III over a range of frequencies, strokes, loading pressures, sealed pressures, and geometries. Values of λ obtained from the above equations varied between 1.6 and 2.4 for all the cases shown in Figures 6-1 through 6-10.

Comparisons of the data for a 50.8-mm stroke at 35 Hz in Figure 6-1 with those for a 25.4-mm stroke at 60 Hz shown in Figure 6-4 indicate that, even though the speeds (product of stroke and frequency) are fairly close together, the short stroke data in Figure 6-4 shows a factor of 3 less flow. This is very much in keeping with the starvation analysis that predicts that reduced flow will occur when the land length becomes an appreciable fraction of the stroke. If starvation is neglected, the theory would predict the flows to be solely a function of the product of frequency and stroke. Even though only a two-constant fit was used, the agreement between theory and experiment for all of the cases involving the small clearance babbitt ring looks reasonably good.

Figure 6-6 shows the comparison between theory and experiment with the larger clearance babbitt ring. The shaded area shows the difference between predictions for a 0.75-mil clearance and a 0.65-mil clearance. The difference being well within the limit of the measurements. In general, the data tend to indicate that the 0.65-mil clearance is probably closer to the truth. This appears to be particularly true when one looks at the data at 10 Hz. Again, the agreement is reasonably good and no additional constants were used in fitting the data for the large clearance ring.

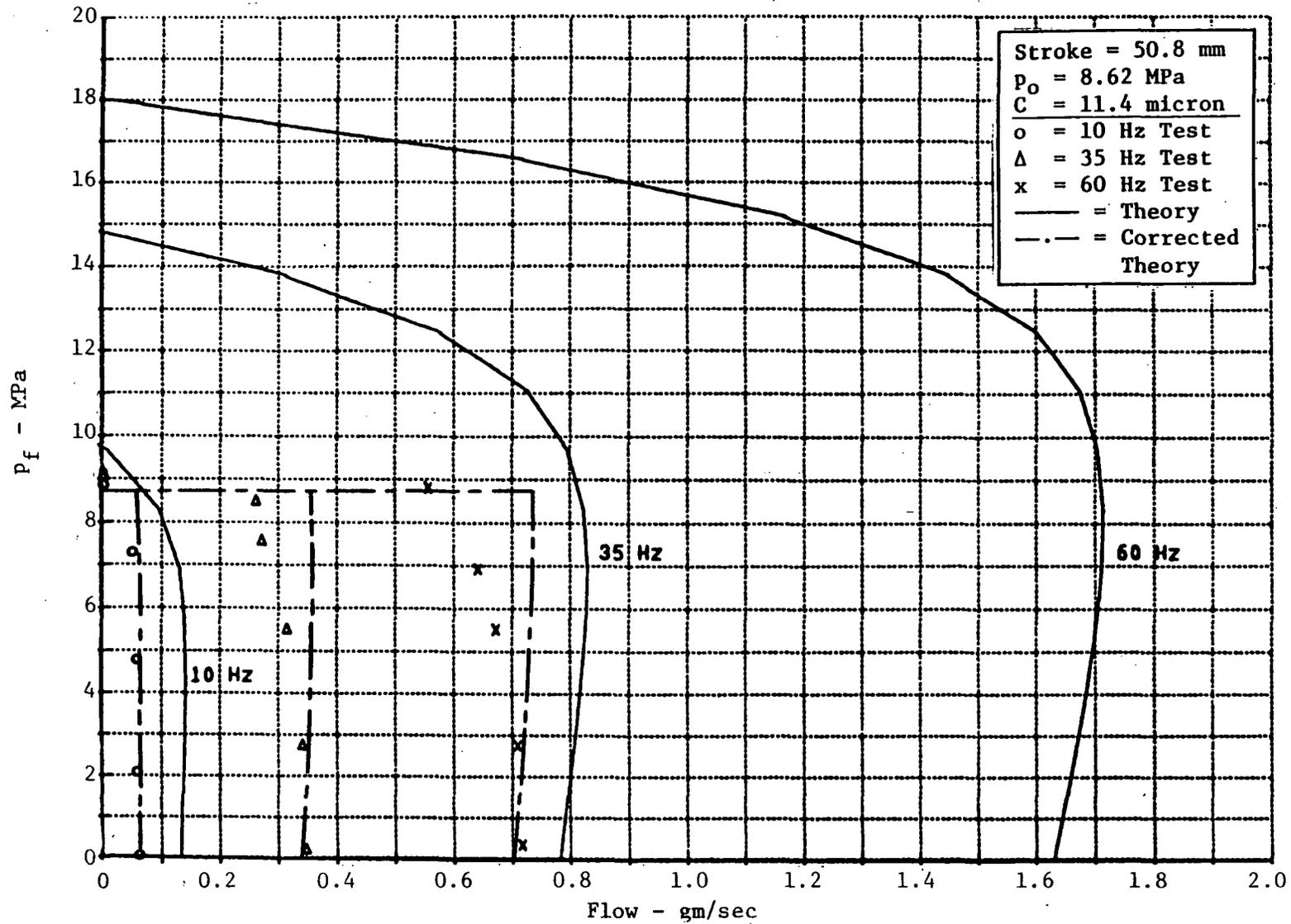


Fig. 6-1 Comparison Between Corrected Theory and Experiment for Babbitt Pumping Ring I
 ($s = 50.8$ mm; $p_0 = 8.62$ MPa)

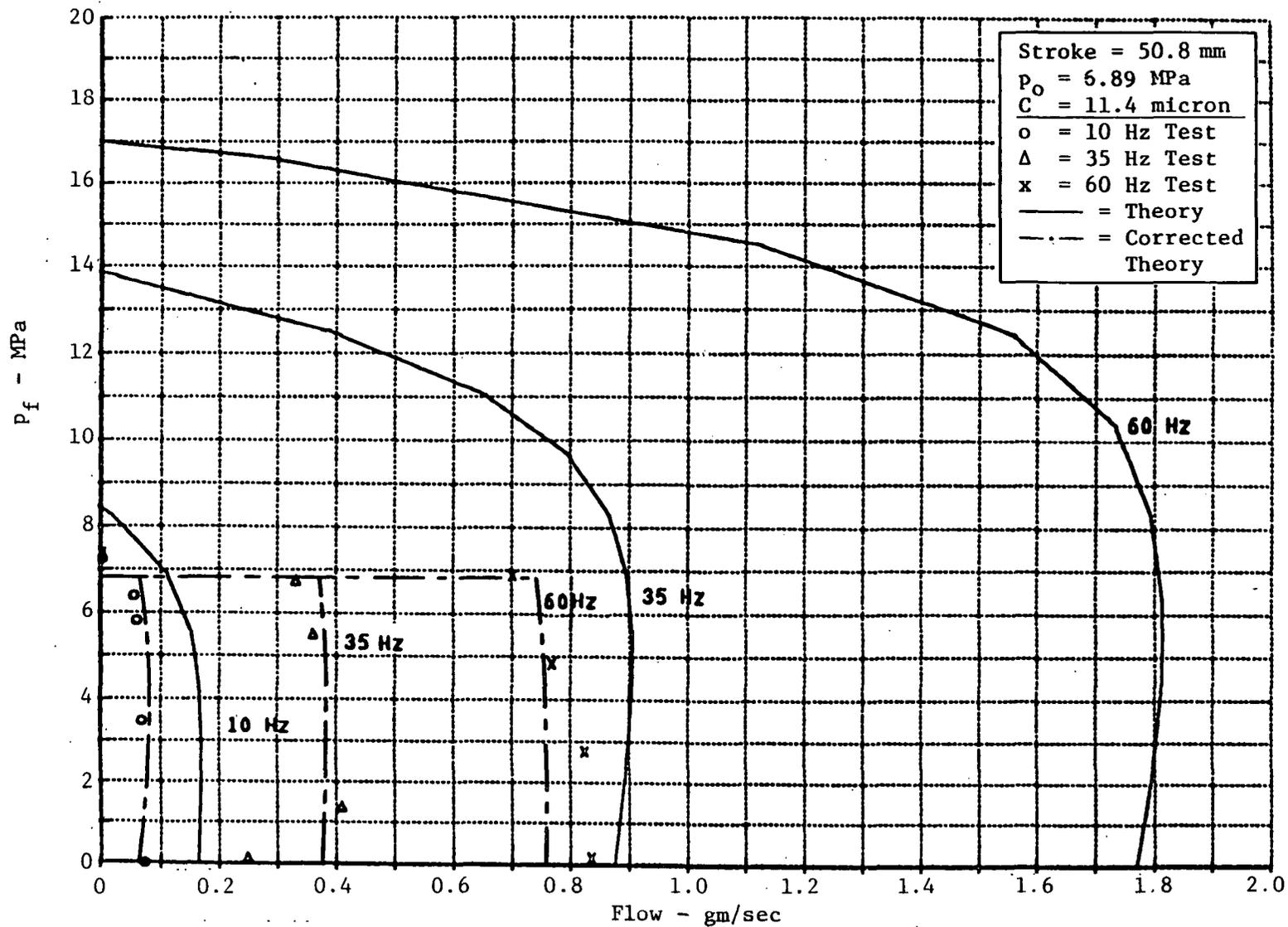


Fig. 6-2 Comparison Between Corrected Theory and Experiment for Babbitt Pumping Ring I
 ($s = 50.8$ mm; $p_0 = 6.89$ MPa)

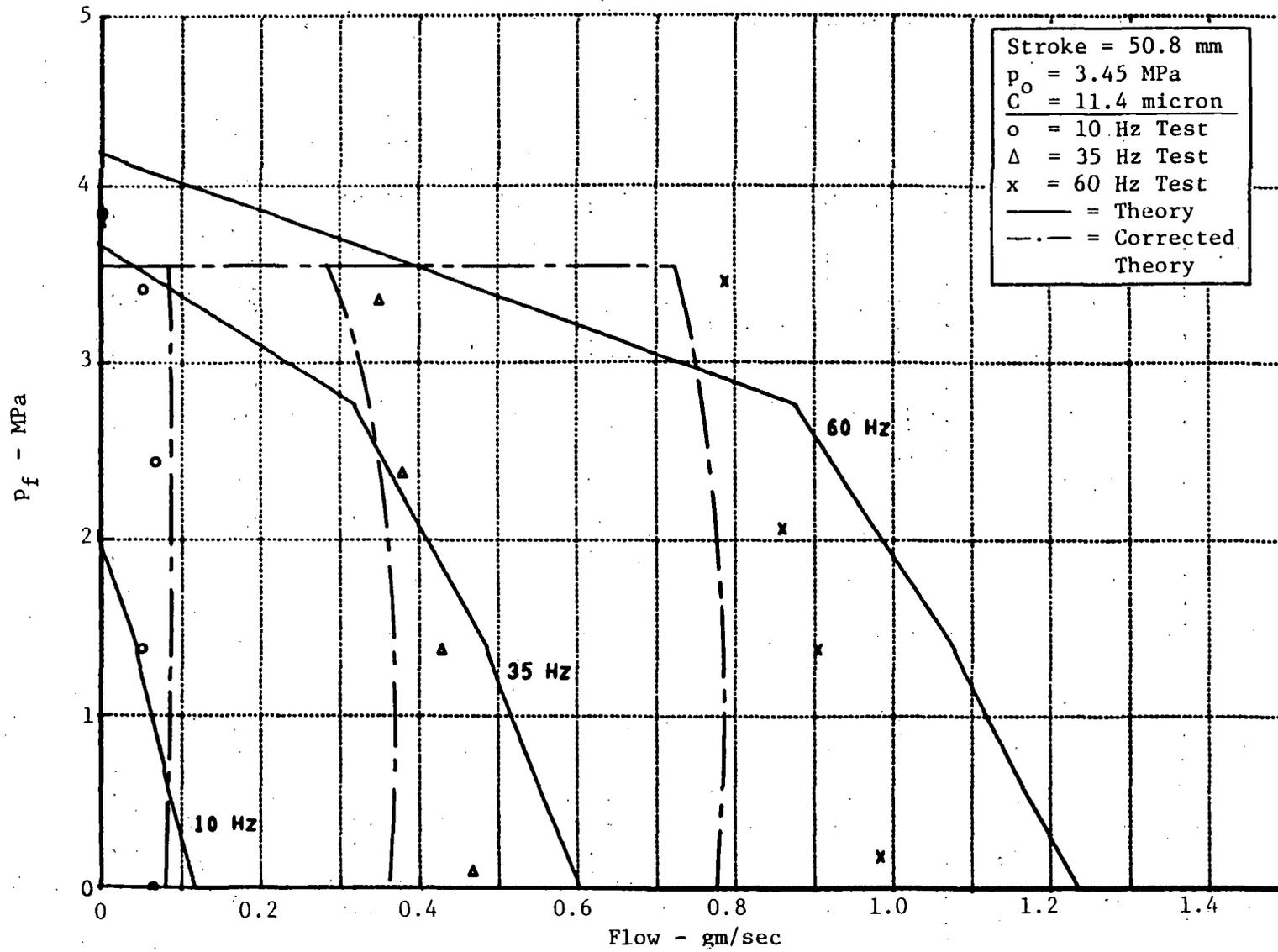


Fig. 6-3 Comparison Between Corrected Theory and Experiment for Babbitt Pumping Ring I
 ($s = 50.8$ mm; $p_0 = 3.45$ MPa)

C-2

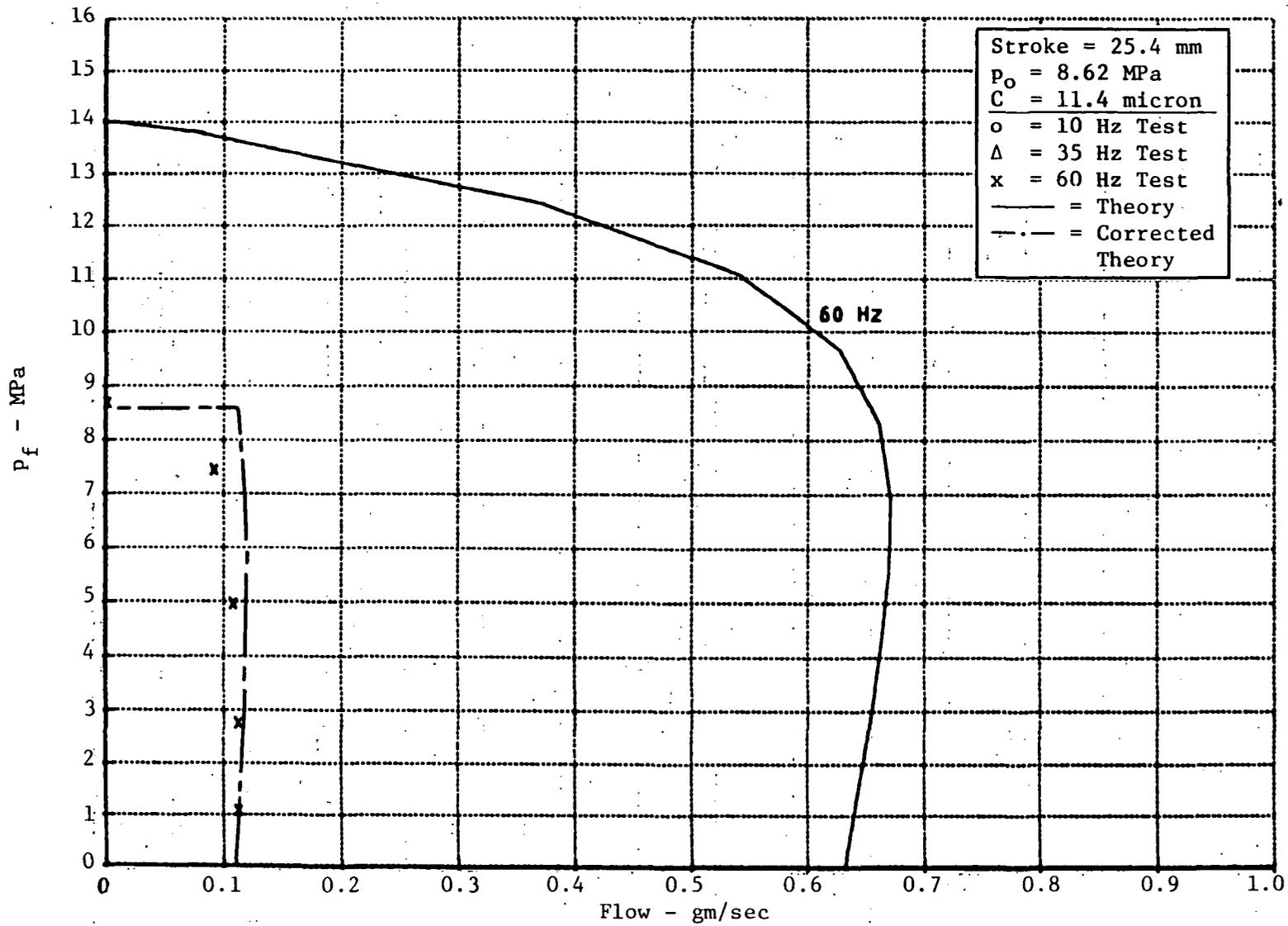


Fig. 6-4 Comparison Between Corrected Theory and Experiment for Babbitt Pumping Ring I
($s = 25.4$ mm; $p_0 = 8.62$ MPa)

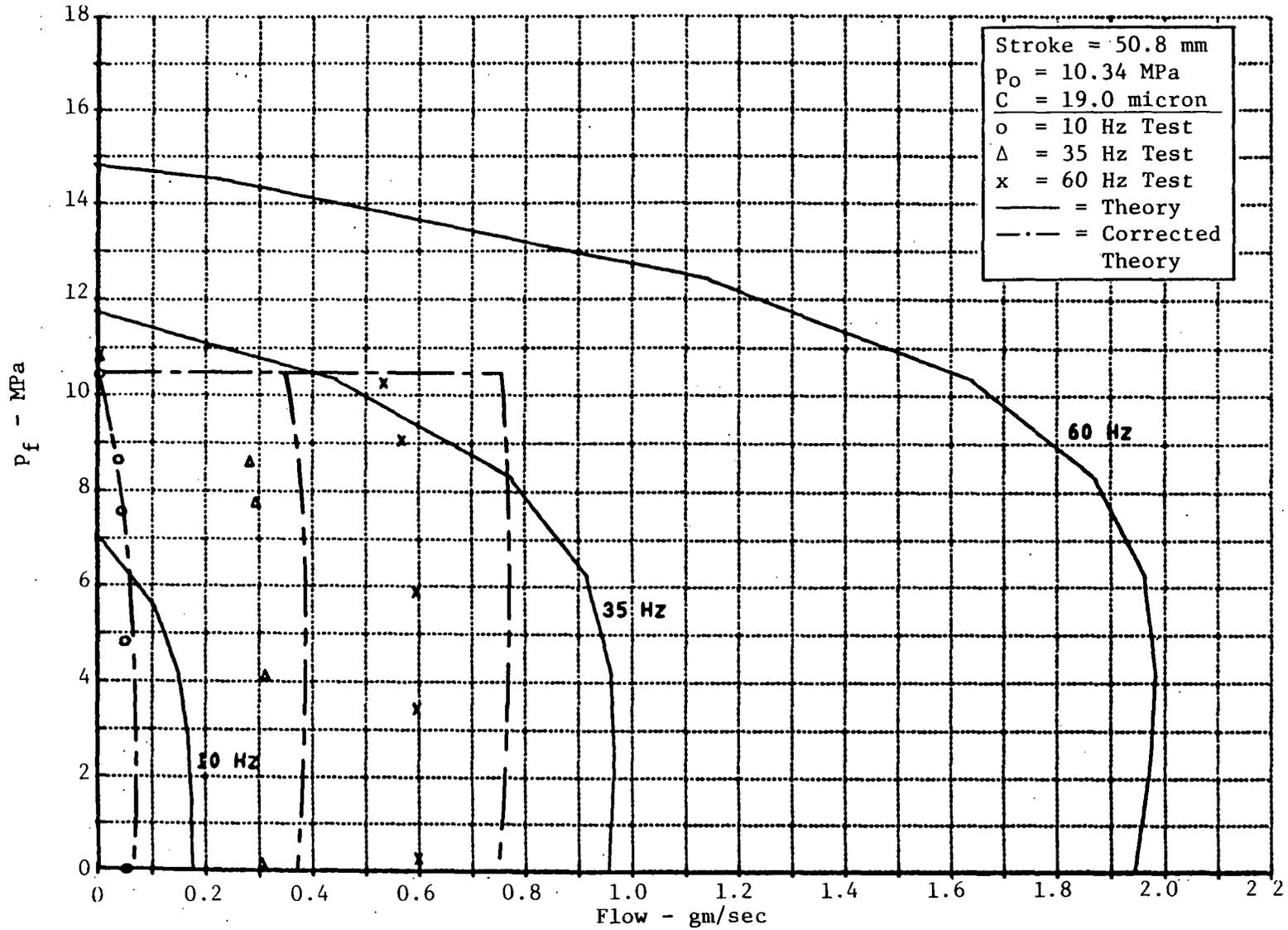


Fig. 6-5 Comparison Between Corrected Theory and Experiment for Babbitt Pumping Ring II
 ($s = 50.8$ mm; $p_0 = 10.34$ MPa)

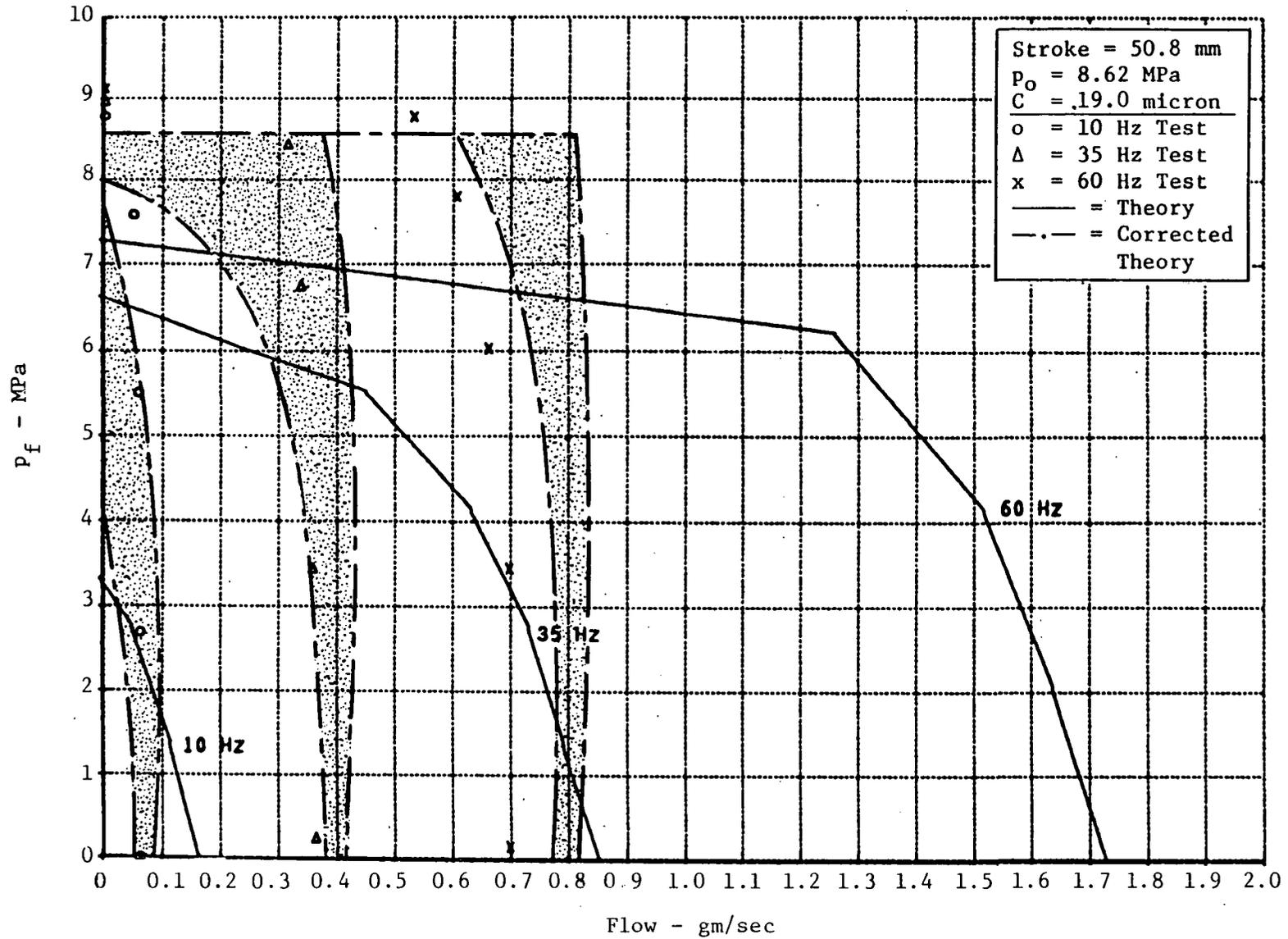


Fig. 6-6 Comparison Between Corrected Theory and Experiment for Babbitt Pumping Ring II.
(s = 50.8 mm; $P_0 = 8.62$ MPa)

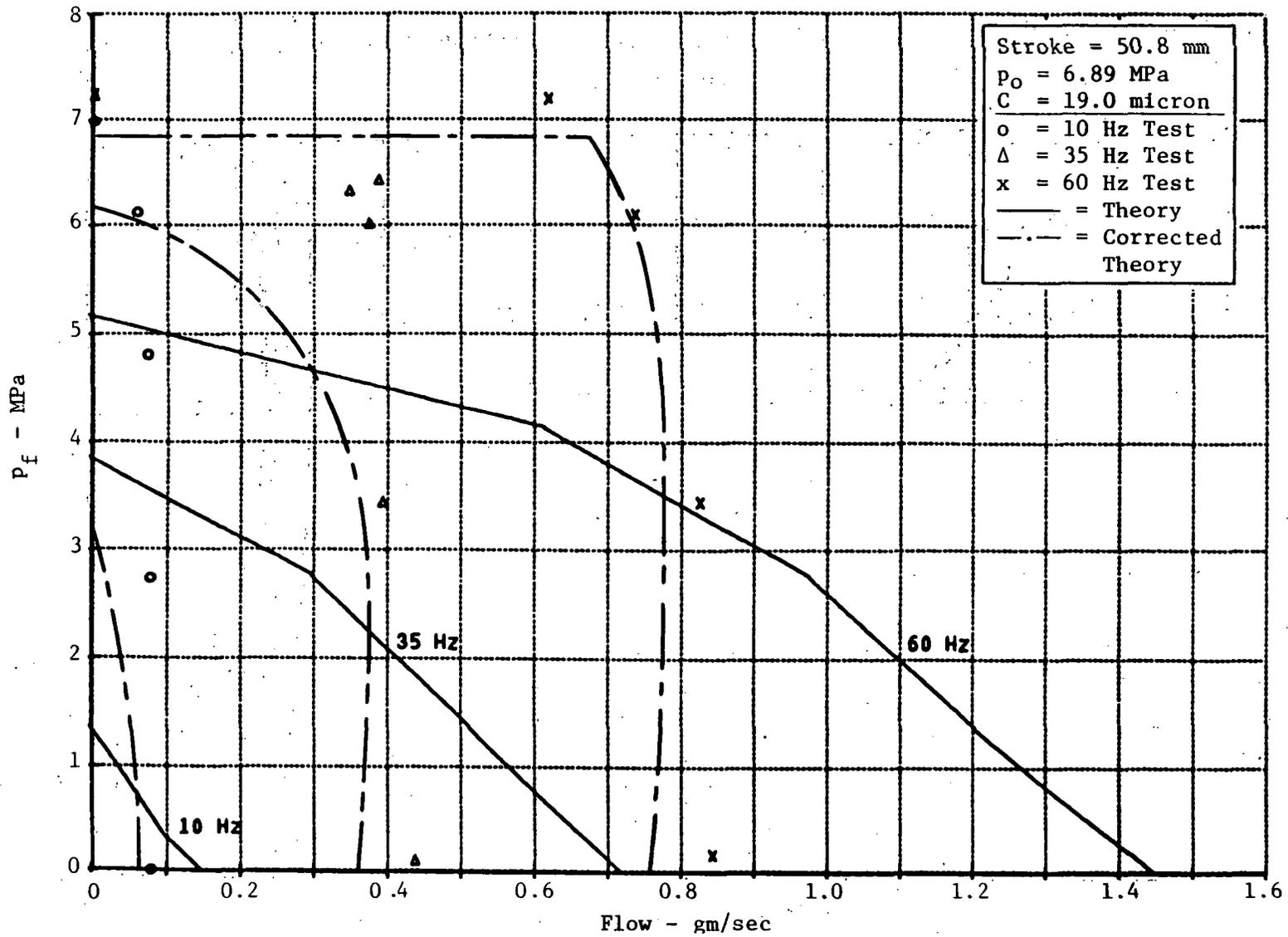


Fig. 6-7 Comparison Between Corrected Theory and Experiment for Babbitt Pumping Ring II.
 ($s = 50.8$ mm; $p_o = 6.89$ MPa)

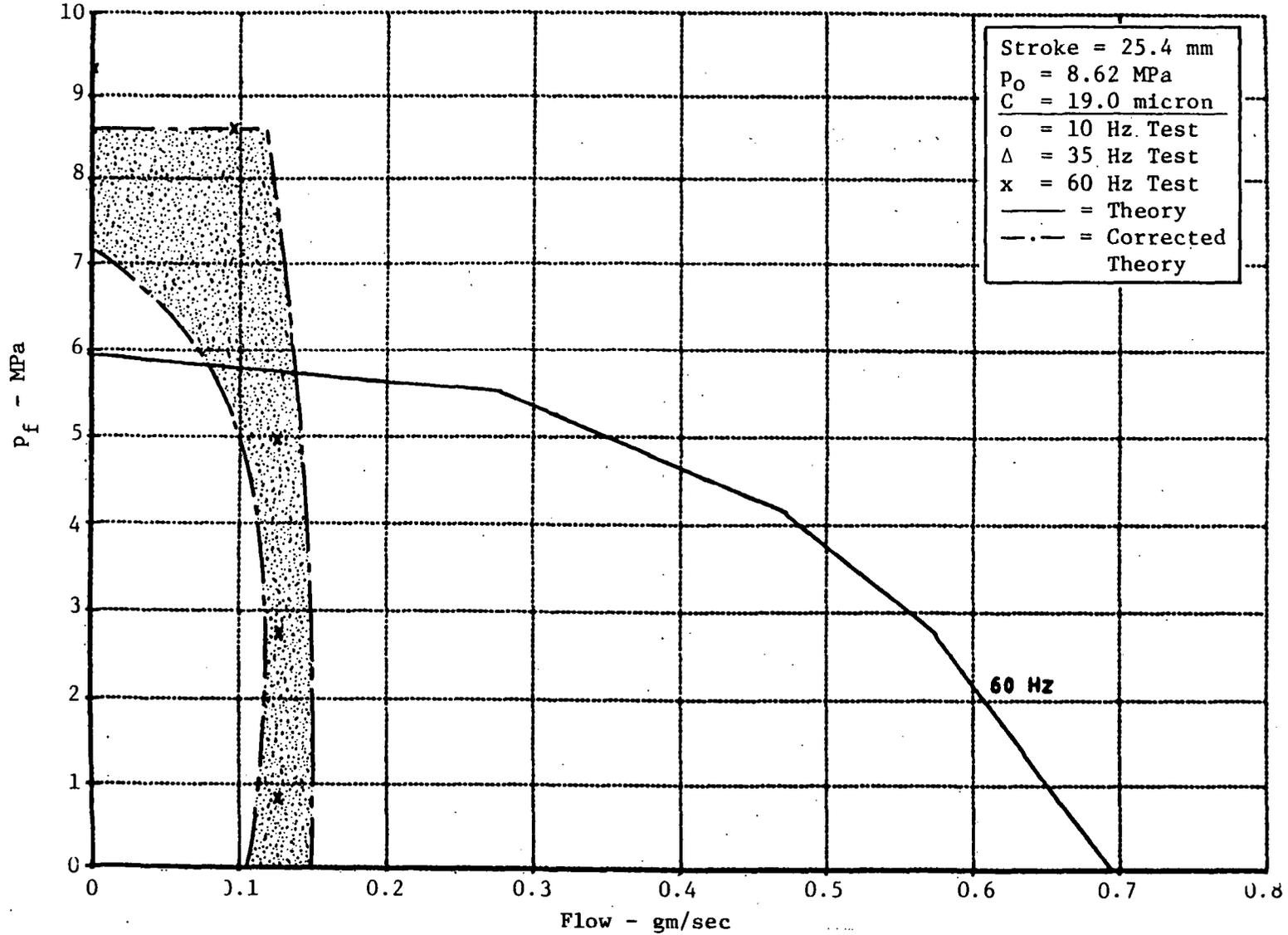


Fig. 6-8 Comparison Between Corrected Theory and Experiment for Babbitt Pumping Ring II
($s = 25.4$ mm; $p_0 = 8.62$ MPa)

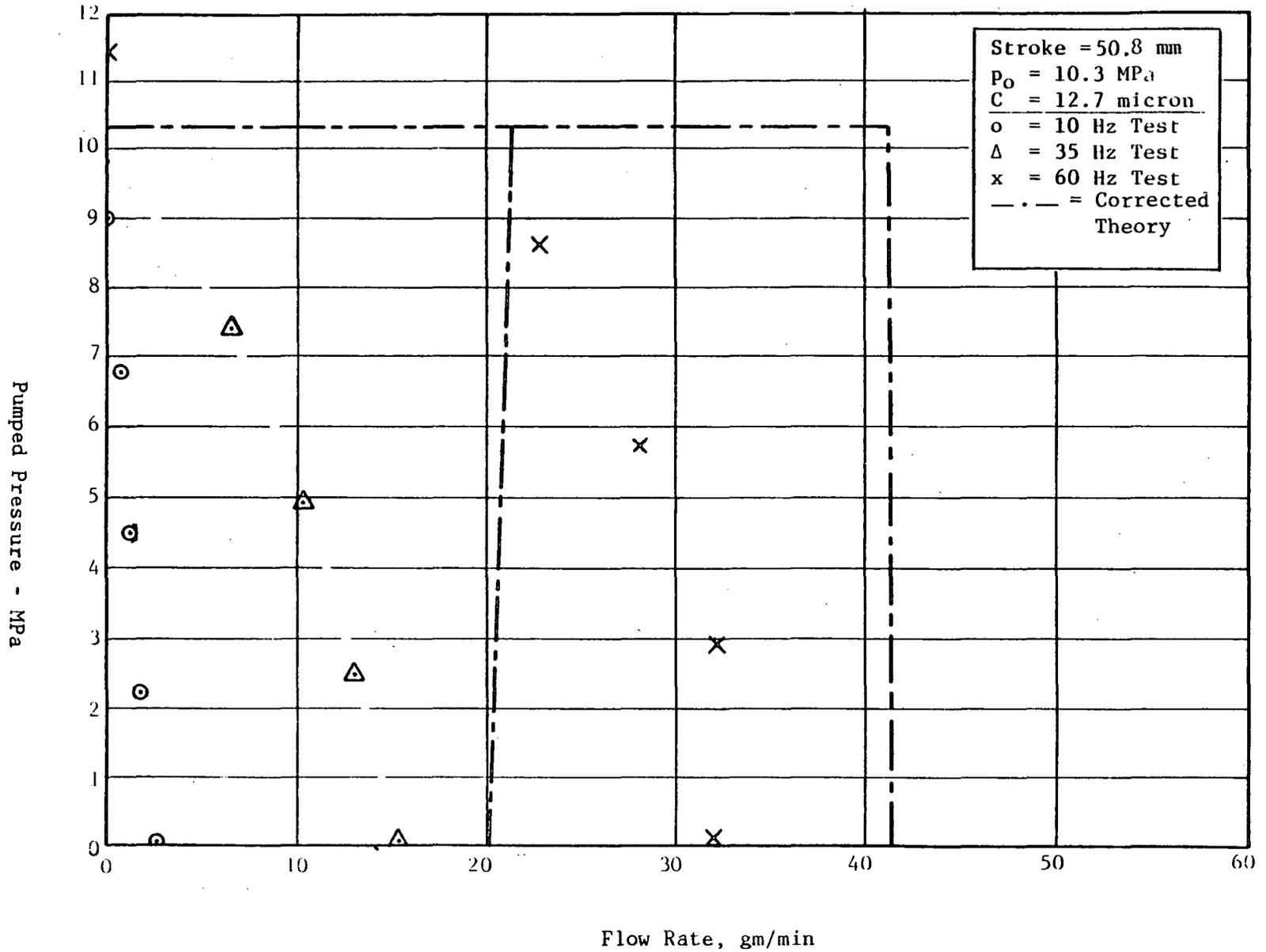


Fig. 6-9 Comparison Between Corrected Theory and Experiment for Babbitt Pumping Ring III
 (s = 50.8 mm; $p_0 = 10.3$ MPa)

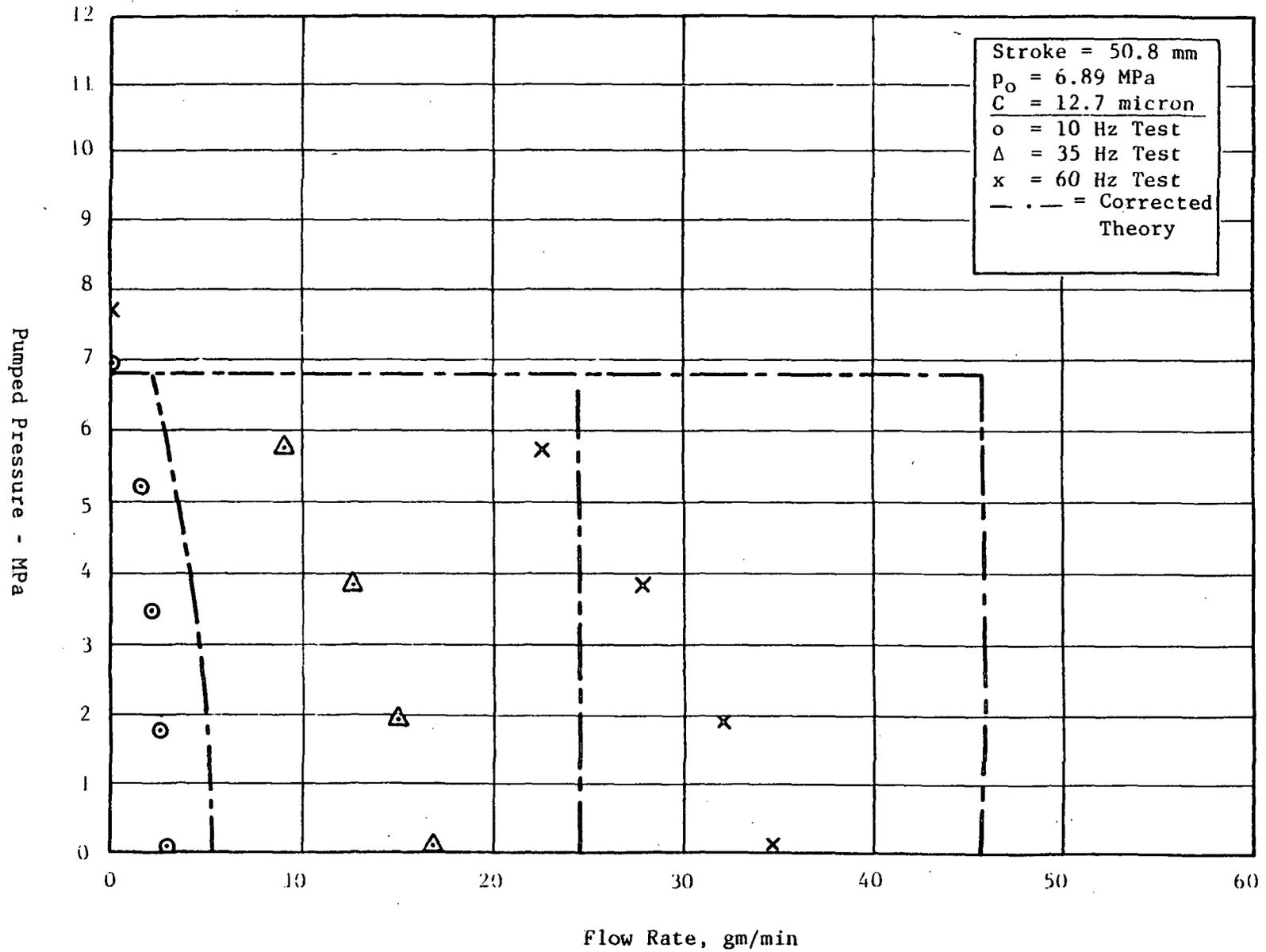


Fig. 6-10 Comparison Between Corrected Theory and Experiment for Babbitt Pumping Ring III
 ($s = 50.8$ mm; $p_0 = 6.89$ MPa)

Figures 6-9 and 6-10 show comparisons between theory and experiment for the shorter length ring. The increased discrepancies indicate that the starvation correction depends more on the stroke than on the seal land. This could be an effect of inertia in the inlet region. In general, the starvation correction, along with the constraint that the end pressure doesn't exceed the loading pressure, provides much better agreement between theory and experiments but is still somewhat incomplete.

6.2 Effects of Viscosity

Since the viscosity cannot be measured at the interface (it is, in fact, measured several inches upstream of the inlet) and since the clearances are very small, both parameters could, in actuality, differ from the assumed values. Major reductions in viscosity, such as that which would occur if the actual temperature were 100°F higher than the measured inlet temperature, would provide remarkably improved agreement between theory and experiment. This fact was observed in numerous parametric studies which are not presented here in that no justification has been found for this temperature rise.

6.3 Suggested Design Procedure

The most effective use of a pumping ring is believed to be for generating a buffer pressure to back up another seal. Thus, if one were to seal gas at 10 MPa from oil at ambient pressure, the pumping ring should be used to develop the 10 MPa backup pressure in a buffer reservoir, thus alleviating the pressure gradient on the primary seal. This should provide prolonged life for the sealing system since the pumping ring is essentially a "light contact" device.

Clearances should be selected based on shaft diameters and machining tolerances. Values of C/R should be of the order of 10^{-3} , thus resembling bearing clearances rather than the tight clearances (or interference) that one would generally find in a seal.

The loading pressure should be of the order of the desired buffer pressure. Back leakage past the O-ring could thus provide a potential relief valve to protect against excessive pressure buildup in the buffer chamber.

The computer program RING (given in Appendix B) should then be used to arrive at a combination of material properties (E, ν) and geometries (t, e, L, L_1) so that the computed value of P_{fm} is equal to the desired buffer pressure, and a small amount of clamping is predicted to occur during the back stroke. Allowances should be made for some wear to occur due to friction resulting from the radial shear force F .

If the pumping ring is designed to generate its own loading pressure, $p_o = p_f$, an initial taper should be used. Values of C_M/C of the order of 2-3 have been shown to be adequate for obtaining self-pumping without the need for external priming. Caution should be used here in providing pressure relief to avoid overloading.

Finally, if the pumping ring is used as a flow device, rather than a pressure generator, allowances should be made for the fact that the analysis tends to overpredict flow under many circumstances. In general, the "starved net flow" prediction generated by RING should be within a factor of 2 of the flow that one would expect to obtain in practice, based upon the measurements reported here.

7.0 CONCLUSIONS

A design analysis has been developed for predicting pressure-flow relationships for various input geometries, speeds, fluid viscosities, and elastic moduli of pumping rings. The analysis can be used to size and design pumping rings for various applications. Results of testing with Rulon, babbitt, and carbon-graphite rings at various loads, speeds, and geometries indicate that the analysis can consistently be used to design rings that work well over a fairly broad range of parameters.

The design criteria for selecting the clearance would be dictated by tolerance and flow rate consideration. The applied loading, p_o , should be the order of the desired pumping pressure. The choice of materials and geometry are then dictated by elasticity considerations so that the deflected ring, under static load, clamps the shaft with a very small force to nearly eliminate the back flow without resulting in excessive friction and ensuing wear and power loss.

The greatest discrepancy between theory and experiment lies in the prediction of flow at zero sealed pressure, Q_o . Predicted values of Q_o are substantially higher than those observed experimentally in almost all cases. Predictions of the maximum sealed pressure at zero net flow, p_{fm} , differ from those observed experimentally, in that values of p_{fm} substantially greater than the loading pressure, p_o , are frequently predicted but have never been observed.

The analysis shows that starvation can have a significant influence on Q_o , although not sufficient to completely explain the discrepancies, and that leakage past the O-ring could cause p_{fm} not to exceed p_o . Further causes of discrepancy could be due to uncertainties in the clearance and the local temperature at the inlet to the pumping ring.

It was found experimentally that the use of tapers on pumping rings enabled the rings to self-pump up to high sealed pressures. The untapered rings, in general, needed to be initially loaded until a sufficiently high sealed pressure could be generated to load the ring. No priming was necessary for the tapered rings. Theory indicates that performance of a tapered ring and an untapered ring at the same minimum film thickness are similar under significant loading, but the tapered ring is predicted to pump even when unloaded.

8.0 REFERENCES

1. "Experimental Evaluation of Oil Pumping Rings," M. W. Eusepi, J. A. Walowit, M. Cohen, DOE/NASA/0119-81/1, NASA CR-165271, U.S. Department of Energy, April 1981.

APPENDIX A

PRELIMINARY ANALYSIS OF PUMPING LENINGRADER SEAL

NOMENCLATURE

δ	Interference
w	Radial inward deflection
h	Film thickness
x_1, x_2	Coordinate variables
p	Interface pressure
p_g	Sealed gas pressure
p_i	Interference pressure
p_o	Ambient oil pressure
E	Elastic modulus of seal
ν	Poisson's ratio
t	Thickness of steel
R	Radius
$D = Et^3/[12(1-\nu^2)]$	Flexural rigidity
$f(x)$	Shape of undeformed seal ring relative to shaft radius
μ	Viscosity
θ	Inlet slope
L, L_1	Lengths shown in Figure A-4

Figure A-1 is a schematic of a pumping Leningrader seal cross section mounted on a shaft. The seal, which is preloaded against the shaft with an interference fit, is loaded by a backup spring and by high-pressure gas, which acts behind the seal and is separated from the oil by the static sealing land. The long, chamfered inlet region provides the pumping action in that it forms a gradual inlet which tends to trap or to provide a preferred direction for flow when the shaft moves toward the cooling supply. (This is referred to as the forward stroke.) During the backstroke, the seal (as drawn in Figure A-1) tends to rub and wipe oil away. Thus, any oil that leaks toward the high-pressure gas tends to be pumped back toward the cooling oil reservoir.

Figure A-2 shows the actual dimensions of the seal used in an automotive Stirling engine. The dotted line denotes the geometry used in the initial modeling of the seal discussed below.

A preliminary model has been developed for a seal with a constant thickness as shown in Figure A-3. The pressure profile drawn under the seal is the pressure anticipated to occur between the seal and the shaft during the forward stroke. The pressure on the far left corresponds to the high-pressure gas, and the pressure on the right represents the oil pressure. The constant pressure, p_i , denotes the compressive radial stress associated with the interference fit of the ring on the shaft. The gas pressure is assumed to act along the outside of the seal with an imaginary secondary seal separating the gas from the oil. The deformation of the seal is assumed to be governed by the elasticity equations for a thin axisymmetric cylindrical shell.

$$D(d^4w/dx^4) + (Et/R^2)w = -(p - p_g) \quad (A-1)$$

The geometric variables and coordinate system are shown in Figure A-4; w denotes the inward radial deflection, E denotes the elastic modulus of the seal, and D denotes the flexural rigidity defined as:

$$D = Et^3/[12(1 - \nu^2)]$$

where ν is Poissons ratio for the seal.

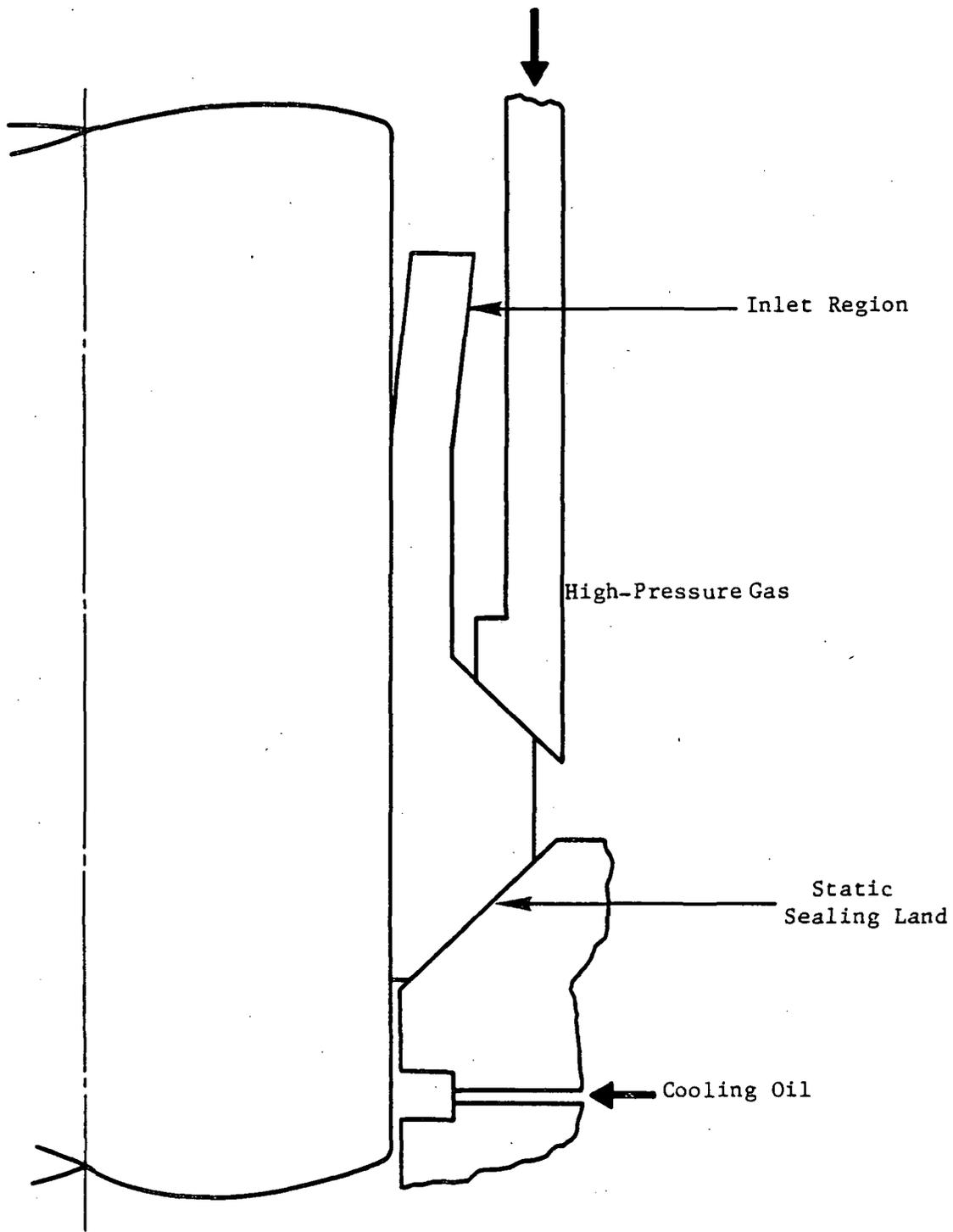


Fig. A-1 Pumping Leningrader Seal

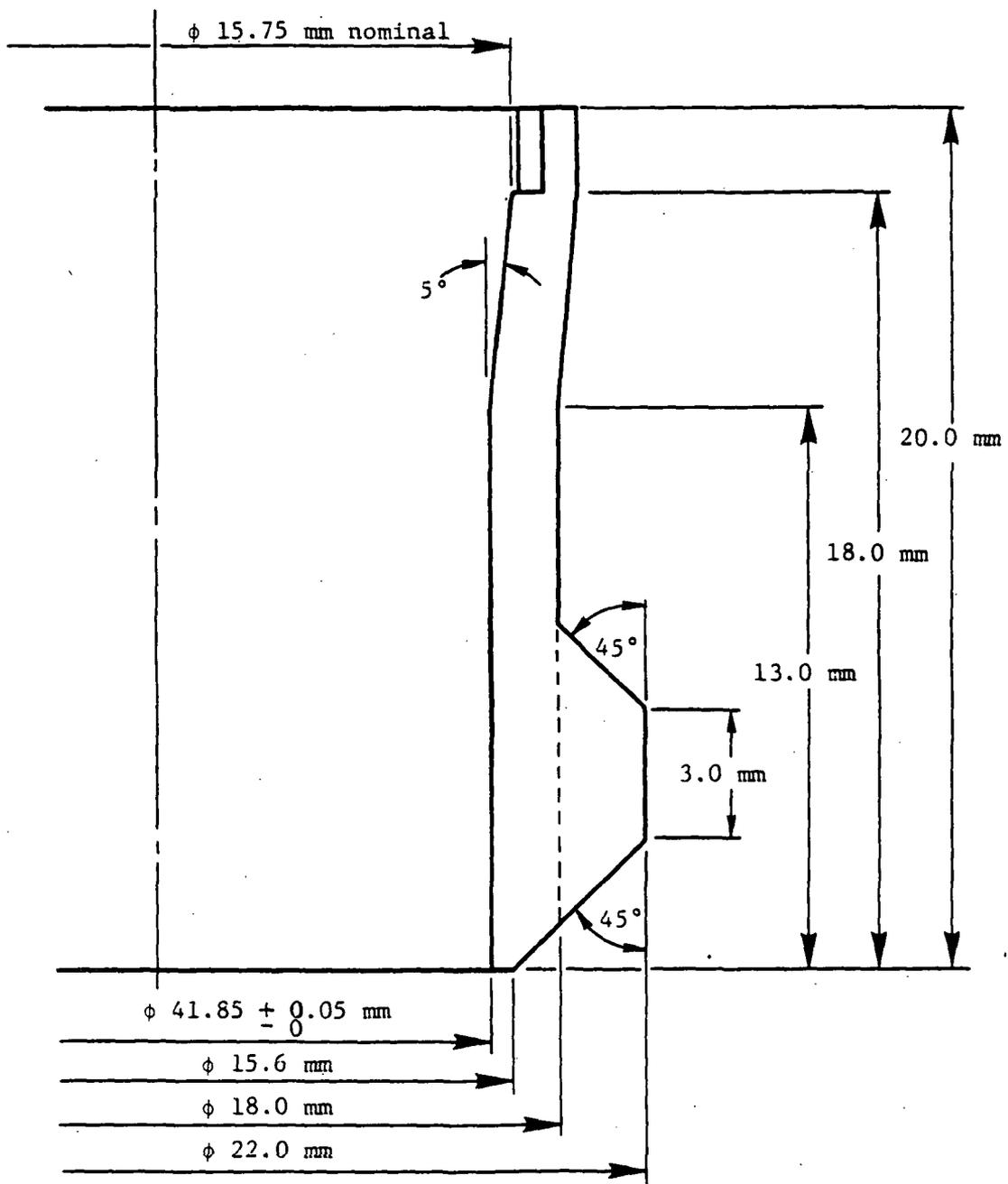


Fig. A-2 Seal Geometry for Automotive Stirling Engine Application

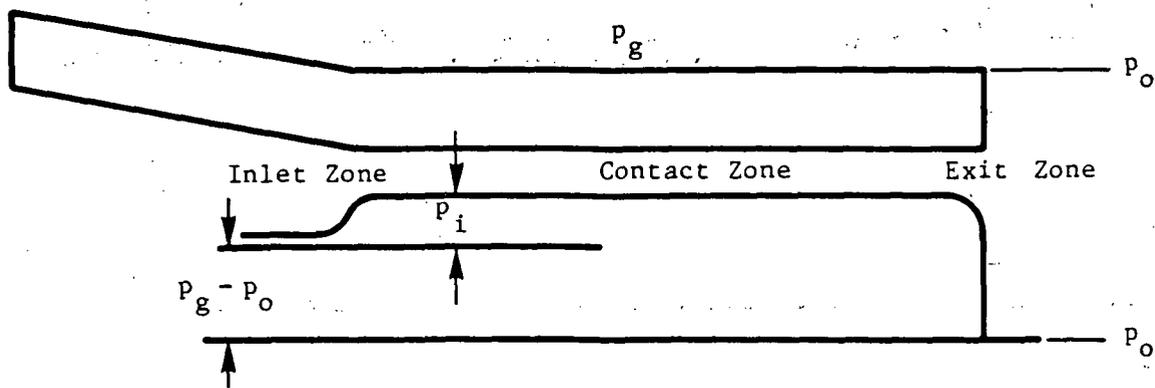


Fig. A-3 Pressures Acting on Pumping Leningrader Seal Used in Preliminary Analysis

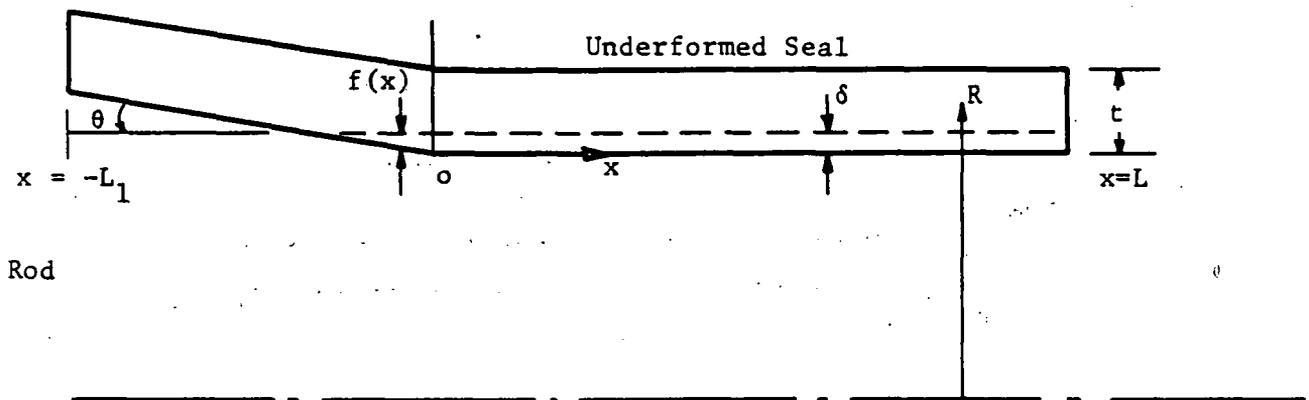


Fig. A-4 Schematic of Geometry Used in the Analysis

The quantity, p , enotes the pressure in the interface between the shaft and the seal which will be determined by the Reynolds Equation for axisymmetric steady-state motion during the forward stroke given by:

$$dp/dx = 6\mu U_0 [(h - h^*)/h^3] \quad (A-2)$$

where h is the local hydrodynamic film thickness, μ is the viscosity of the oil, U_0 is the average velocity during the forward stroke, and h^* is a constant of integration to be determined from the boundary conditions.

The film thickness h is related to the elastic deflection w by:

$$h = f(x) - w \quad (A-3)$$

where $f(x)$ is the radius of the undeformed seal relative to the shaft radius. For the model depicted in Figure A-4, $f(x)$ is given by:

$$f(x) = \begin{cases} -\delta & , \quad 0 \leq x \leq L \\ -\delta - \theta x, & -L_1 \leq x \leq 0 \end{cases} \quad (A-4)$$

The boundary conditions on the elasticity equation come from the requirement that both ends of the seal are free (forces and moments are zero at each end of the seal). The boundary conditions on the Reynolds equation comes from the requirements that the hydrodynamic pressures equal the prescribed gas pressure at the high-pressure side and the prescribed oil pressure on the low-pressure side. These may be written as:

$$d^2w/dx^2 = d^3w/dx^3 = 0 \text{ at } x = -L_1 \text{ and } x = L \quad (A-5)$$

$$p = p_g \text{ at } x = -L_1 \quad (A-6)$$

$$p = p_o \text{ at } x = L \quad (A-7)$$

Equations (A-1, through (A-7) represent a fifth-order nonlinear system corresponding to the model under consideration.

A number of approximations that are characteristic of elastohydrodynamic lubrication may be introduced. These are based largely on the fact that the hydrodynamic film thickness, h , will be small compared with the interference fit, δ . The elastohydrodynamic behavior may be characterized by three zones:

- an inlet zone where the film develops
- a contact zone where the film thickness is nearly constant
- an exit zone where the pressure drops to ambient.

This type of behavior has been seen in many other types of elastohydrodynamic lubrication, e.g., Hertzian contacts in rolling element bearings, elastomeric bearings and foil bearings.

A.1 Contact Zone Solution

The contact zone is the region extending from $x = 0$ to the start of the exit zone which will be localized near $x = L$. In that region, from Equation (A-4), $f(x) = -\delta$ and from Equation (A-3), $w = -\delta - h$, which may be approximated by $w = \delta$ since $|h/\delta| \ll 1$. With w being constant, the left-hand term in Equation (A-1) will vanish; hence the relationship

$$p - p_g \equiv p_i = (Et\delta)/R^2, \text{ contact zone pressure} \quad (\text{A-8})$$

Since p as determined in Equation (A-8) is constant in the contact zone, the left-hand term in Equation (A-2) will vanish and the result will be $h = h^*$, constant (contact zone film thickness).

Equation (A-8) is essentially the dry contact pressure resulting from the interference fit. The dry contact profile would also contain a radial shear load at the sharp corner at $x = 0$. With a small amount of wear, however, this corner will be rounded, thus alleviating the radial shear load without radically affecting the inlet film shape. Equation (A-8) will then be assumed to prevail throughout the contact zone with the value of h^* to be determined by matching the pressure obtained from the solution to the equations for the inlet zone.

A.2 Inlet Zone Solution

In order to match the solution in the inlet zone with the contact zone solution, the pressure and flow must be continuous at $x = 0$. This results in the constraint

$$p = p_g + p_i \text{ at } x = 0 \quad (\text{A-9})$$

and

$$dp/dx = 0 \text{ at } x = 0$$

which is equivalent to

$$h = h^* \text{ at } x = 0 \quad (\text{A-10})$$

For values of θ of the order of a few degrees or more, the pressure will vary from its limiting values of $p_g + p_i$ at $x = 0$ to p_g over a relatively small portion of the inlet zone as sketched in Figure A-3. If the shape of the inlet zone is assumed to be unaffected by the pressure distribution in the region, the film thickness relationship is

$$h = h^* - \theta x, \quad x < 0$$

The stretched variable $\xi = -\theta x/h^*$ may now be introduced into Equation (A-2) to obtain

$$dp/d\xi = -6\mu U_0 \xi / [h^*(1 + \xi)^3] \quad (\text{A-11})$$

with the constraints

$$p = p_g + p_i \text{ at } \xi = 0 \quad (\text{A-12})$$

$$\text{and } p = p_g \text{ at } \xi = L_1\theta/h^* \quad (\text{A-13})$$

The quantity $L_1\theta/h^*$ will be a large number for practical cases. Equation (A-13) may be replaced by its limiting form

$$p = p_g \text{ at } \xi \rightarrow \infty \quad (A-14)$$

Equation (A-11) may readily be integrated to yield

$$p_i = \frac{6\mu U_o}{\theta h^*} \int_0^\infty \frac{\xi}{(1+\xi)^3} d\xi = 3\mu U_o / (\theta h^*)$$

Thus the film thickness h^* is given by -

$$h^* = 3\mu U_o / (\theta p_i) = 3\mu U_o R^2 / (Et\delta\theta) \quad (A-15)$$

which may, in turn, be used to calculate flow.

A.3 Exit Zone Solution

The exit zone solution is the region in the neighborhood of $x = L$ where the pressure must drop from $p_g + p_i$ down to the ambient value of p_o . This will, again, occur over a relatively short region. The film thickness deflection relationship for the exit zone as determined from Equations (A-3) and (A-4) is the same as that for the contact zone and may be written as

$$w = -(h + \delta)$$

The above relationship may be substituted for w in Equation (A-1), to obtain

$$D(d^4 h/dx_1^4) + Et(\delta + h)/R^2 = p - p_g$$

where $x_1 = L - x$.

Neglecting h compared with δ in the second term of the above equation and employing Equation (A-8), one obtains

$$D(d^4h/dx_1^4) = p - (p_g + p_i) \quad (A-16)$$

One may now differentiate Equation (A-16) with respect to x_1 and substitute Equation (A-2) for $-dp/dx_1$ to obtain the exit zone equation

$$D(d^5h/dx_1^5) = -6\mu U_o (h - h^*)/h^3 \quad (A-17)$$

where h^* is now determined from Equation (A-15).

The boundary conditions at $x_1 = 0$ may be obtained from Equations (A-5), (A-7) and (A-16)

$$d^2h/dx_1^2 = d^3h/dx_1^3 = 0 \text{ at } x_1 = 0 \quad (A-18)$$

and

$$D(d^4h/dx_1^4) = p_o - p_g - p_i \text{ at } x = 0 \quad (A-19)$$

Without going through the formality of introducing a stretching transformation as was done for the inlet zone, the exit zone solution must merge with the contact zone solution as $x_1 \rightarrow \infty$. Thus, we look for a bounded solution with $h \rightarrow h^*$ as $x_1 \rightarrow \infty$.

The above system of equations may be solved numerically by Runge-Kutta integration, however; if the variation in h is not excessive ($|h - h^*| \ll h^*$), then one may replace h^3 appearing in the denominator of the right-hand side of Equation (A-17) with h^{*3} to obtain a linear system of equations with constant coefficients that may readily be solved algebraically.

The algebraic solution is in the form:

$$h/h^* = 1 + A_1 e^{-\zeta} + e^{-\lambda_1 \zeta} (A_2 \cos \lambda_2 \zeta + A_3 \sin \lambda_2 \zeta)$$

where

$$\zeta = [6\mu U_o / (h^{*3} D)]^{1/5} x_1, \quad \lambda_1 = \cos(2\pi/5) \text{ and } \lambda_2 = \sin(2\pi/5).$$

The constants A_1 , A_2 and A_3 are readily determined from Equations (A-18) and (A-19).

A.4 Sample Solution

Film thickness and pressure profiles have been calculated corresponding to the geometry shown in Figure A-2 with the trapezoidal section modeled by the dotted line shown in the figure. A fluid viscosity of 55 cp, an elastic modulus of 1.72 GPa, and a Poisson's ratio of 0.41 have been used in the computation.

As shown on the scale in Figure A-5, the inlet zone film thickness profile is very steep, indicating that only the portion of the inlet zone very near the contact zone is relevant in affecting the film thickness profile, and that the major portion of the seal to the left of this region is probably not necessary. This is again shown in Figure A-6, where the pressure, approximately 1 mm before the start of the contact zone, is very nearly that of the gas (10 MPa). The pressure rises very steeply and joins the pressure in the contact zone smoothly, although it appears to have a sharp corner on the scale in Figure A-6. The pressure profile in the contact zone is constant and equal to the sum of the gas pressure plus the interference pressure (the pressure that would be associated with interference fit statically). Near the exit region, the pressure starts to oscillate, peaks to 18 MPa, and then falls rapidly to 0 (the oil pressure).

Examination of the rapid falloff in pressure at the exit region shows that if the exit oil pressure were raised to be equal to, or even greater than, the gas pressure, there would be little change in the pressure profile except in the immediate vicinity of the exit zone. The film thickness in the contact zone would be virtually unaffected. This indicates that the ambient pressure difference has little effect on the film thickness or the flow rate since the flow is completely dominated by viscous forces. Thus, the seal could be made to provide a film thickness in either direction by adding a chamfer on the reverse side. Such chamfers, which were recommended and tested as part of the Automotive Stirling Engine Program, have been shown to be effective in providing lubrication without excessive leakage in either direction.

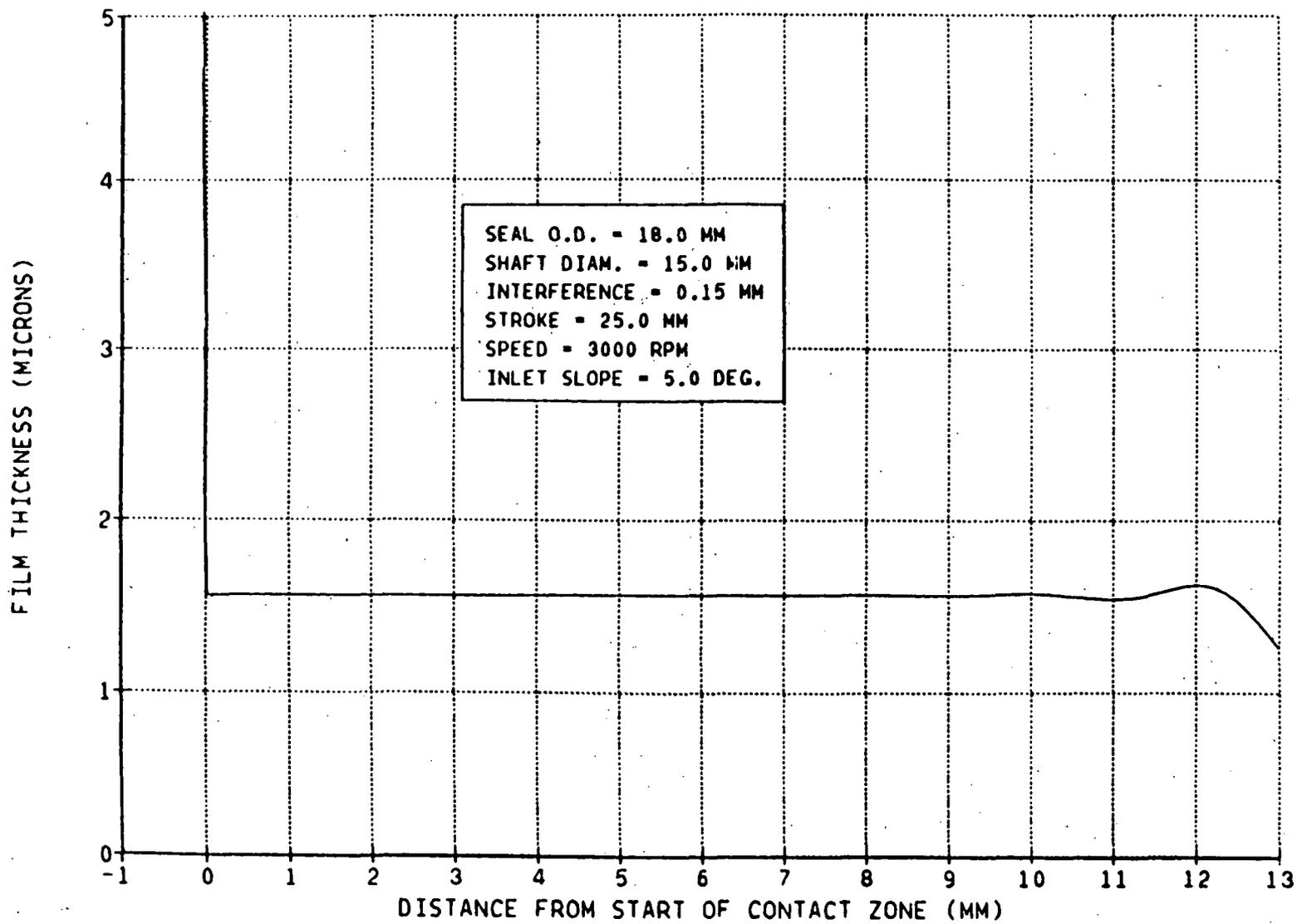


Fig. A-5 Film Thickness Profile

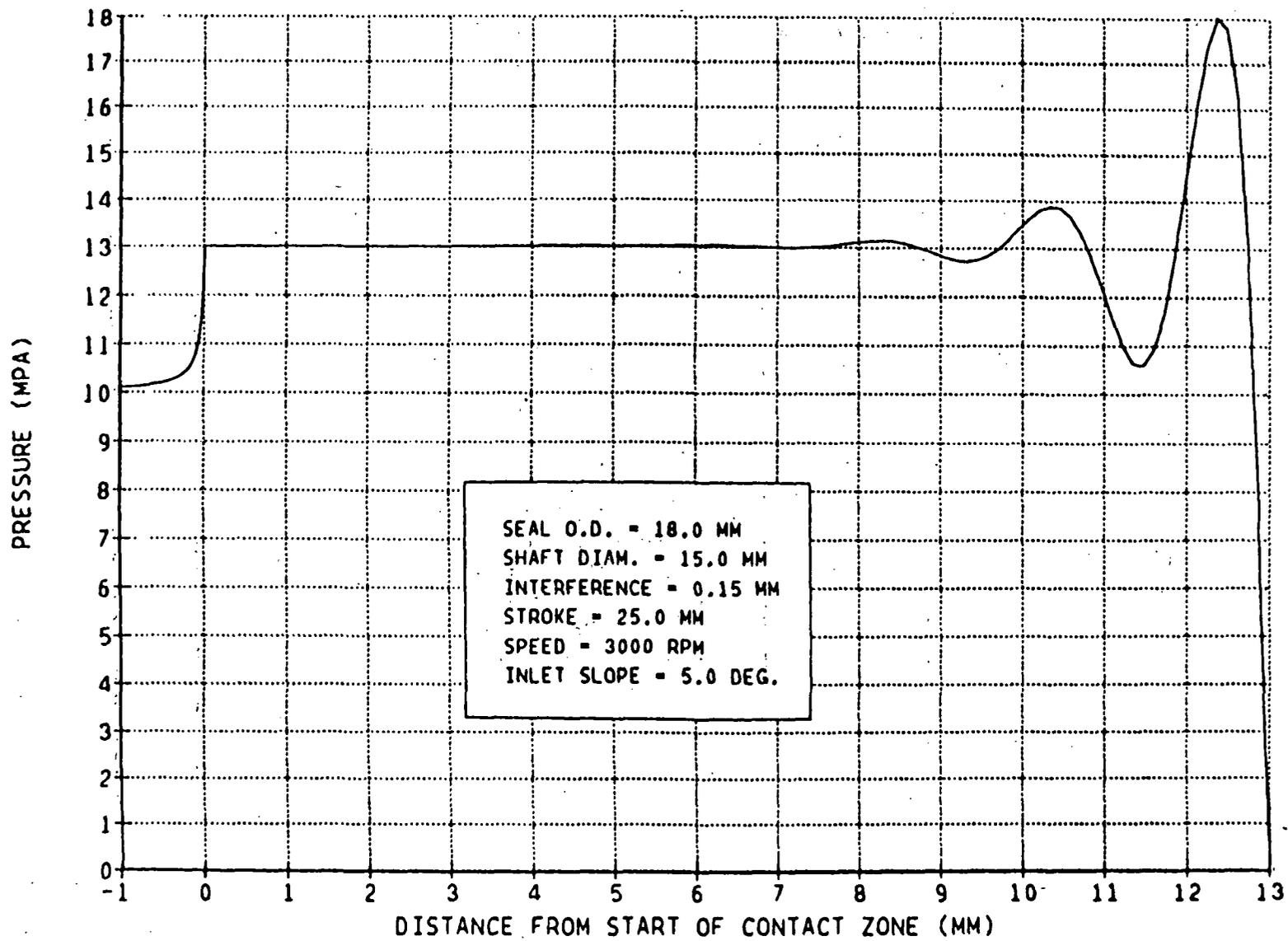


Fig. A-6. Pressure Profile

APPENDIX B

COMPUTER PROGRAM "RING"

The computer program RING which has been used throughout the report is listed herein. It is written in IBM FORTRAN 77 (VS), and uses an IMSL subroutine ZSCNT. A FORTRAN IV listing of ZSCNT and its subroutines is also included for completeness. The FORTRAN listings are preceded with an input description, an output description, and sample inputs and outputs.

B.1 INPUT DESCRIPTION

All input variables are on a namelist (NAMELIST/INPUTS/). Symbols in the left column denote the FORTRAN names of the namelist variables. Symbols appearing on the right correspond to the nomenclature given at the beginning of the report.

Input Variable Definitions

STRING	A character string of up to 60 characters to identify job
IDIM	Type of input
	0 - Dimensionless (default value)
	1 - Dimensional (any consistent set of units)

Dimensional Inputs (used only if IDIM = 1)

NI	Frequency (Hz), f
SI	Stroke (L), s
CI	Clearance (L), c
ELI	Bearing length (L), L
ELII	Length of nonbearing portion of ring (L), L_1
EI	Length over which p_0 acts (L), e
RI	Shaft radius (L), R
TI	Ring thickness (L), t
CLI	Initial taper, δ (defaults to 0)
POI	Preload pressure (F/L^2), p_0
PFI	Reservoir pressure (F/L^2), p_f
RMUI	Lubricant viscosity (FT/L^2), μ
EMOD	Ring modulus of elasticity (F/L^2), E
POIS	Ring Poisson's Ratio, ν

Nondimensional Inputs

AL	α
BET	β
EPS	ϵ

EL1 L₁
 C1 $\delta L/C$ (defaults to 0)
 P0 P₀
 PF P_f
 ELOS L/s for starvation calculation (defaults to 0)

NDX A vector of length 3 containing number of increments for Runge-Kutta integration
 NDX(1) = Number of increments in $1 - \epsilon \leq \xi \leq 1$ (defaults to 40)
 NDX(2) = Number of increments in $0 \leq \xi \leq 1 - \epsilon$ (defaults to 25)
 NDX(3) = Number of increments in $-L_1 \leq \xi \leq 0$ (defaults to 25)

NDP A vector of length (3) containing number of points at which pressures etc. are saved for printout. NDX(i) should be integral multiples of NDP(i). (Defaults are NDP(1) = 10, NDP(2) = 25, NDP(3) = 25.)

ICAV Flag calculation of cavitation (backstroke)
 0 - No cavitation (negative pressures allowed)
 2 - Cavitation included, no negative pressures (default value)

IPR Print Flag
 0 - Short output (default value)
 1 - Longer output (print pressure profile etc.)

NMAX Maximum iterations for secant method solution (default is 10)

NSIG Number of significant digits in accuracy of solution using secant method (default is 4)

XF A vector of length (2) containing initial guesses for forward stroke secant solution
 XF(1) = Initial guess for h at $\xi = 1$
 XF(2) = Initial guess for dh/d ξ at $\xi = 1$

If XF(1) and XF(2) are both set equal to 0, the program will attempt to find its own initial guesses. In subsequent runs, values determined from the previous run are used unless explicitly specified.

XB A vector length (2) containing initial guesses for h and $dh/d\xi$ for reverse stroke secant solution. Same rules apply to XB as for XF.

B.2 OUTPUT DESCRIPTION

The output starts with a listing of the inputs which are defined in D.1. Dimensional inputs are listed only if IDIM = 1. All outputs are dimensionless.

The Elastic Solution refers to the deflection of the ring when hydrodynamic forces are not present. If the ring is unclamped, $h(1) > 0$, one line of output is given for the Elastic Solution. If the ring is clamped, $h(1) < 0$, two lines of output are given. The first line refers to the ring deflection relative to an imaginary shaft that would occur if the shaft were not present (the negative value of h noting the interference). The second line of output includes the effect of the interference force exerted by the shaft on the ring at $\xi = 1$ when $h(1) = 0$.

The Pumping Flow Solution and Back Flow Solution refer to forward and reverse strokes respectively. They will contain one line of output each at IPR = 0 or multiple outputs if IPR = 1.

The following table relates the output caption with the symbols for variables as defined in the nomenclature for the elastic, pumping flow and back flow solution.

<u>Output Caption</u>	<u>Algebraic Symbol</u>
X	ξ
H	h
H'	$dh/d\xi$
H''	$d^2h/d\xi^2$
H'''	$d^3h/d\xi^3$
PRES	P
FORCE	F

The final output listed under flows relates to dimensionless flow and cavitation output and are given as follows:

Output Caption

Algebraic Symbol

PUMPING

K_F (Forward Stroke)

BACK

K_R (Reverse Stroke)

XCAV

ξ_c

NET FLOW

$K_F - K_R$

STARVED NET FLOW

$K_F \text{ eff} - K_R$

B.3 SAMPLE INPUT AND OUTPUT

The sample input and output contained herein correspond to three solutions for a babbitt pumping ring with loading pressures, p_0 , of 3.45, 5.17 and 6.89 MPa (500, 750 and 1000 psi). A full printout (IPR = 1) was requested at $p_0 = 3.45$ MPa (500 psi). It should be noted that the ring is predicted to be clamped during the backstroke at $p_0 = 6.89$ MPa (1000 psi).

&INPUTS

STRING='BABBIT RING A-1-A-1 PO=500 COMPLETE OUTPUT (IPR=1)'

IDIM=1, NI =35., SI = 2., CII =0.,
CI =.5E-3, RMUI =.885E-5, ELI =.267, POI =500.,
PFI =0., EI =.115, ELII =.298, RI =.375,
TI =.047, EMOD =7.5E6, POIS =.36,
XF = 0.,0., XB = 0.,0., IPR=1,

&END

&INPUTS POI=750.,STRING ='BABBIT RING A-1-A-1 PO=750',IPR=0,&END

&INPUTS POI=1000.,STRING ='BABBIT RING A-1-A-1 PO=1000',&END

* PUMPING RING ANALYSIS PROGRAM *

INPUTS

 STRING = BABBIT RING A-1-A-1 PO=500 COMPLETE OUTPUT (IPR=1)

DIMENSIONAL INPUTS

ELI = 0.26700000D+00; ELII = 0.29800000D+00;
 EI = 0.11500000D+00; CI = 0.50000000D-03;
 CII = 0.00000000D+00; TI = 0.47000000D-01;
 SI = 0.20000000D+01; RI = 0.37500000D+00;
 RMUI = 0.88500000D-05; NI = 0.35000000D+02;

POI = 0.50000000D+03; PFI = 0.00000000D+00;
 EMOD = 0.75000000D+07; POIS = 0.36000000D+00;

IDIM = 1;
 AL = 0.66085654D-02; BET = 0.71535454D+01;
 EPS = 0.43071161D+00; PO = 0.62976163D-01;
 EL1 = 0.11161049D+01; PF = 0.00000000D+00;
 STARV = 0.32041966D+00; C1 = 0.00000000D+00;

NDX = 40, 25, 25;
 NDP = 10, 25, 25;

NMAX = 10; NSIG = 4;
 IPR = 1; ICAV = 2;
 XF = 0.00000D+00 0.00000D+00
 XB = 0.00000D+00 0.00000D+00

OUTPUTS

* ELASTIC SOLUTION *

X	H	H'	H''	H'''	PRES	FORCE
1.0000	0.48845	-0.67281	0.00000	0.00000	0.00000	0.00000

* PUMPING FLOW SOLUTION *

X	H	H'	H''	H'''	PRES
1.0000	0.81814	-0.69559	0.00000	0.00000	0.00000
0.9914	0.82413	-0.69558	-0.00147	0.33645	0.00375
0.9742	0.83611	-0.69548	-0.01254	0.93187	0.01074
0.9569	0.84809	-0.69510	-0.03304	1.43418	0.01709
0.9397	0.86006	-0.69429	-0.06149	1.85487	0.02285
0.9225	0.87201	-0.69294	-0.09655	2.20463	0.02805
0.9052	0.88393	-0.69093	-0.13710	2.49334	0.03274
0.8880	0.89582	-0.68819	-0.18216	2.73019	0.03695
0.8708	0.90764	-0.68464	-0.23092	2.92369	0.04071
0.8536	0.91940	-0.68022	-0.28270	3.08173	0.04405
0.8363	0.93108	-0.67488	-0.33695	3.21161	0.04701
0.8191	0.94265	-0.66859	-0.39324	3.32008	0.04960
0.8019	0.95411	-0.66132	-0.45126	3.41339	0.05186

0.7846	0.96543	-0.65304	-0.51080	3.49729	0.05380
0.7674	0.97660	-0.64371	-0.57175	3.57709	0.05545
0.7502	0.98761	-0.63333	-0.63406	3.65768	0.05682
0.7330	0.99842	-0.62186	-0.69781	3.74354	0.05793
0.7157	1.00903	-0.60927	-0.76311	3.83880	0.05881
0.6985	1.01941	-0.59555	-0.83016	3.94719	0.05946
0.6813	1.02954	-0.58066	-0.89921	4.07216	0.05990
0.6640	1.03941	-0.56456	-0.97058	4.21679	0.06015
0.6468	1.04899	-0.54720	-1.04464	4.38389	0.06021
0.6296	1.05825	-0.52854	-1.12178	4.57598	0.06011
0.6124	1.06719	-0.50853	-1.20247	4.79530	0.05984
0.5951	1.07577	-0.48709	-1.28718	5.04382	0.05943
0.5779	1.08396	-0.46415	-1.37644	5.32328	0.05887
0.5693	1.08791	-0.45209	-1.42295	5.47508	0.05855
0.5579	1.09297	-0.43555	-1.48207	4.91228	0.05807
0.5351	1.10249	-0.40062	-1.58150	3.83132	0.05695
0.5124	1.11120	-0.36370	-1.65702	2.81167	0.05564
0.4896	1.11904	-0.32532	-1.71003	1.85497	0.05416
0.4668	1.12600	-0.28598	-1.74199	0.96235	0.05252
0.4440	1.13206	-0.24614	-1.75435	0.13448	0.05073
0.4213	1.13721	-0.20622	-1.74861	-0.62831	0.04881
0.3985	1.14146	-0.16663	-1.72623	-1.32603	0.04676
0.3757	1.14481	-0.12772	-1.68871	-1.95894	0.04461
0.3530	1.14728	-0.08982	-1.63751	-2.52748	0.04236
0.3302	1.14891	-0.05323	-1.57408	-3.03230	0.04001
0.3074	1.14972	-0.01821	-1.49988	-3.47412	0.03759
0.2846	1.14975	0.01500	-1.41633	-3.85377	0.03509
0.2619	1.14905	0.04623	-1.32484	-4.17212	0.03252
0.2391	1.14766	0.07529	-1.22678	-4.43005	0.02989
0.2163	1.14564	0.10206	-1.12353	-4.62844	0.02721
0.1936	1.14303	0.12643	-1.01643	-4.76813	0.02448
0.1708	1.13990	0.14833	-0.90681	-4.84992	0.02171
0.1480	1.13630	0.16772	-0.79599	-4.87453	0.01890
0.1252	1.13228	0.18459	-0.68524	-4.84260	0.01606
0.1025	1.12791	0.19894	-0.57586	-4.75471	0.01319
0.0797	1.12324	0.21083	-0.46912	-4.61129	0.01029
0.0569	1.11833	0.22034	-0.36627	-4.41272	0.00737
0.0342	1.11322	0.22755	-0.26857	-4.15924	0.00443
0.0114	1.10798	0.23262	-0.17726	-3.85102	0.00148
-0.0000	1.10532	0.23439	-0.13439	-3.67640	0.00000

* BACK FLOW SOLUTION *

X	H	H'	H''	H'''	PRES
1.0000	0.48845	-0.67281	0.00000	0.00000	0.00000
0.9914	0.49424	-0.67281	0.00033	-0.07580	0.00000
0.9742	0.50584	-0.67284	0.00279	-0.20473	0.00000
0.9569	0.51743	-0.67292	0.00721	-0.30344	0.00000
0.9397	0.52902	-0.67309	0.01307	-0.37193	0.00000
0.9225	0.54062	-0.67338	0.01985	-0.41019	0.00000
0.9052	0.55223	-0.67378	0.02703	-0.41820	0.00000
0.8880	0.56384	-0.67431	0.03409	-0.39595	0.00000
0.8708	0.57546	-0.67495	0.04050	-0.34341	0.00000
0.8536	0.58710	-0.67570	0.04575	-0.26055	0.00000
0.8363	0.59875	-0.67652	0.04931	-0.14735	0.00000

0.8191	0.61041	-0.67738	0.05065	-0.00376	0.00000
0.8019	0.62209	-0.67825	0.04926	0.17025	0.00000
0.7846	0.63378	-0.67906	0.04461	0.37472	0.00000
0.7674	0.64548	-0.67976	0.03617	0.60970	0.00000
0.7502	0.65720	-0.68028	0.02343	0.87520	0.00000
0.7330	0.66892	-0.68054	0.00584	1.17126	0.00000
0.7157	0.68065	-0.68045	-0.01711	1.49788	0.00000
0.6985	0.69237	-0.67992	-0.04595	1.85507	0.00000
0.6813	0.70407	-0.67883	-0.08120	2.24279	0.00000
0.6640	0.71575	-0.67708	-0.12340	2.66099	0.00000
0.6468	0.72740	-0.67454	-0.17307	3.10961	0.00000
0.6296	0.73899	-0.67107	-0.23072	3.58851	0.00000
0.6124	0.75051	-0.66654	-0.29689	4.09756	0.00000
0.5951	0.76195	-0.66079	-0.37208	4.63653	0.00000
0.5779	0.77327	-0.65366	-0.45682	5.20518	0.00000
0.5693	0.77889	-0.64953	-0.50292	5.50054	0.00000
0.5579	0.78625	-0.64346	-0.56341	5.12594	0.00000
0.5351	0.80075	-0.62936	-0.67194	4.41448	0.00000
0.5124	0.81490	-0.61297	-0.76483	3.75238	0.00000
0.4896	0.82865	-0.59464	-0.84320	3.13837	0.00000
0.4668	0.84196	-0.57467	-0.90812	2.57100	0.00000
0.4440	0.85481	-0.55337	-0.96063	2.04872	0.00000
0.4213	0.86716	-0.53101	-1.00175	1.56985	0.00000
0.3985	0.87899	-0.50783	-1.03244	1.13265	0.00000
0.3757	0.89028	-0.48406	-1.05364	0.73528	0.00000
0.3530	0.90103	-0.45991	-1.06622	0.37590	0.00000
0.3302	0.91123	-0.43556	-1.07103	0.05260	0.00000
0.3074	0.92087	-0.41118	-1.06887	-0.23653	0.00000
0.2846	0.92995	-0.38693	-1.06050	-0.49338	0.00000
0.2619	0.93849	-0.36293	-1.04663	-0.71988	0.00000
0.2391	0.94649	-0.33930	-1.02793	-0.91790	0.00000
0.2163	0.95395	-0.31614	-1.00503	-1.08929	0.00000
0.1936	0.96089	-0.29355	-0.97851	-1.23587	0.00000
0.1708	0.96732	-0.27160	-0.94892	-1.35941	0.00000
0.1480	0.97326	-0.25036	-0.91676	-1.46164	0.00000
0.1252	0.97873	-0.22987	-0.88250	-1.54421	0.00000
0.1025	0.98374	-0.21018	-0.84657	-1.60875	0.00000
0.0797	0.98831	-0.19132	-0.80936	-1.65679	0.00000
0.0569	0.99246	-0.17332	-0.77123	-1.68981	0.00000
0.0342	0.99621	-0.15620	-0.73251	-1.70922	0.00000
0.0114	0.99958	-0.13996	-0.69348	-1.71637	0.00000
-0.0000	1.00113	-0.13218	-0.67394	-1.71575	0.00000

 * FLOWS *

PUMPING	BACK	XCAV	NET FLOW	STARVED NET FLOW
0.106219D+01	0.488604D+00	0.999770D+00	0.573582D+00	0.296125D+00

* PUMPING RING ANALYSIS PROGRAM *

INPUTS

STRING = BABBIT RING A-1-A-1 PO=750

DIMENSIONAL INPUTS

ELI = 0.26700000D+00; ELII = 0.29800000D+00;
 EI = 0.11500000D+00; CI = 0.50000000D-03;
 CII = 0.00000000D+00; TI = 0.47000000D-01;
 SI = 0.20000000D+01; RI = 0.37500000D+00;
 RMUI = 0.88500000D-05; NI = 0.35000000D+02;

POI = 0.75000000D+03; PFI = 0.00000000D+00;
 EMOD = 0.75000000D+07; POIS = 0.36000000D+00;

IDIM = 1;
 AL = 0.66085654D-02; BET = 0.71535454D+01;
 EPS = 0.43071161D+00; PO = 0.94464244D-01;
 EL1 = 0.11161049D+01; PF = 0.00000000D+00;
 STARV = 0.32041966D+00; C1 = 0.00000000D+00;

NDX = 40, 25, 25;
 NDP = 10, 25, 25;

NMAX = 10; NSIG = 4;
 IPR = 0; ICAV = 2;
 XF = 0.81814D+00 -0.69559D+00
 XB = 0.48845D+00 -0.67281D+00

OUTPUTS

 * ELASTIC SOLUTION *

X	H	H'	H''	H'''	PRES	FORCE
1.0000	0.23267	-1.00922	0.00000	0.00000	0.00000	0.00000

 * PUMPING FLOW SOLUTION *

X	H	H'	H''	H'''	PRES
1.0000	0.71873	-0.93560	0.00000	0.00000	0.00000

 * BACK FLOW SOLUTION *

X	H	H'	H''	H'''	PRES
1.0000	0.23267	-1.00922	0.00000	0.00000	0.00000

 * FLOWS *

PUMPING	BACK	XCAV	NET FLOW	STARVED NET FLOW
0.100210D+01	0.232747D+00	0.999927D+00	0.769350D+00	0.329915D+00

* PUMPING RING ANALYSIS PROGRAM *

INPUTS

STRING = BABBIT RING A-1-A-1 PO=1000

DIMENSIONAL INPUTS

ELI = 0.26700000D+00; EL1I = 0.29800000D+00;
 EI = 0.11500000D+00; CI = 0.50000000D-03;
 C1I = 0.00000000D+00; TI = 0.47000000D-01;
 SI = 0.20000000D+01; RI = 0.37500000D+00;
 RMUI = 0.88500000D-05; NI = 0.35000000D+02;

POI = 0.10000000D+04; PFI = 0.00000000D+00;
 EMOD = 0.75000000D+07; POIS = 0.36000000D+00;

IDIM = 1;
 AL = 0.66085654D-02; BET = 0.71535454D+01;
 EPS = 0.43071161D+00; PO = 0.12595233D+00;
 EL1 = 0.11161049D+01; PF = 0.00000000D+00;
 STARV = 0.32041966D+00; C1 = 0.00000000D+00;

NDX = 40, 25, 25;
 NDP = 10, 25, 25;

NMAX = 10; NSIG = 4;
 IPR = 0; ICAV = 2;
 XF = 0.71873D+00 -0.93560D+00
 XB = 0.23267D+00 -0.10092D+01

OUTPUTS

* ELASTIC SOLUTION *

X	H	H'	H''	H'''	PRES	FORCE
1.0000	-0.02310	-1.34562	0.00000	0.00000	0.00000	0.00000
1.0000	0.00000	-1.28833	0.00000	-0.70479	0.00000	0.00065

* PUMPING FLOW SOLUTION *

X	H	H'	H''	H'''	PRES
1.0000	0.63082	-1.13187	0.00000	0.00000	0.00000

* BACK FLOW SOLUTION *

X	H	H'	H''	H'''	PRES	FORCE
1.0000	0.00000	-1.28833	0.00000	-0.70479	0.00000	0.00065

* FLOWS *

PUMPING	BACK	XCAV	NET FLOW	STARVED NET FLOW
0.929129D+00	0.000000D+00	0.100000D+01	0.929129D+00	0.333707D+00

B.4 FORTRAN LISTING

FILE: RING FORTRAN L4

MTI

MON 01/20/86 14:39:26

PAGE 1 OF 38

```

PROGRAM RING
C$ VS
C>-----RIN00010
CNOTE: XLAM=.739*ELOS**-.585 IS NOW BUILT INTO PROGRAM RIN00020
C RIN00030
C RIN00040
C RIN00050
C RIN00060
C FUNCTION - PUMPING RING ANALYSIS PROGRAM RIN00070
C RIN00080
C RESTRICTIONS - STEADY-STATE ANALYSIS ONLY RIN00090
C RIN00100
C REMARKS - VS FORTRAN (FORTRAN 77) RIN00110
C IMPLICIT DOUBLE PRECISION REAL*8 RIN00120
C RIN00130
C EXTERNAL REFERENCES - RIN00140
C FORTRAN ROUTINES RIN00150
C DATAN RIN00160
C IMSL ROUTINES RIN00170
C ZSCNT ; SOLVES THE SYSTEM OF NON-LINEAR RIN00180
C EQUATIONS ; LISTING PROVIDED HEREIN BY RIN00190
C PERMISSION OF IMSL. RIN00200
C USER ROUTINES RIN00210
C AIN ; COMPUTE INVERSE OF 2X2 MATRIX RIN00220
C AMU ; MULTIPLY 2 2X2 MATRICES RIN00230
C CALCD ; CALCULATE ELASTIC INFLUENCE RIN00240
C COEFFICIENTS C,D RIN00250
C CHECK ; FOR MULTIPLE RUNS CHECK INPUTS RIN00260
C FOR RECALCULATION OF C,D RIN00270
C CONST ; CALCULATE SLIDER BEARING PRESSURE RIN00280
C CONSTANTS RIN00290
C CONST2 ; SAME AS ABOVE RIN00300
C DFN1 ; DERIVATIVE FUNCTION USED BY RUK RIN00310
C DFN2 ; " " " " " RIN00320
C DFN3 ; " " " " " RIN00330
C ELAS ; DETERMINE ELASTIC SOLUTION RIN00340
C (NO HYDRODYNAMICS) RIN00350
C ERRMSG ; PRINT ERROR MESSAGE IF *ZSCNT* NOT RIN00360
C CONVERGED RIN00370
C EVAL ; DEFINE NON-LINEAR SYSTEM IN H AND H' RIN00380
C TO BE SOLVED BY *ZSCNT* RIN00390
C P ; PRESSURE FUNCTION RIN00400
C PRT ; CONVERT FROM W TO H AND PRINT RIN00410
C PRTOUT ; PRINT OUT RESULTS RIN00420
C RUK ; RUNGE-KUTTA RIN00430
C RIN00440
C INPUT/OUTPUT: RIN00450
C UNIT DESCRIPTION RIN00460
C 4 TERMINAL I/O RIN00470
C 5 INPUT FILE IN NAMELIST FORMAT RIN00480
C 6 OUTPUT FILE RIN00490
C RIN00500
C INPUT VARIABLE DEFINITIONS RIN00510
C * NOTE : (D) INDICATES VARIABLE HAS A DEFAULT VALUE RIN00520
C RIN00530
C NAME DESCRIPTION RIN00540
C NAMELIST /INPUTS/ RIN00550

```

```

C
C          STRING      CHARACTER STRING TO IDENTIFY JOB (MAX. 60 CHARS.) RIN00560
C          IDIM (D)    TYPE OF INPUT RIN00570
C                    0 - DIMENSIONLESS RIN00580
C                    1 - DIMENSIONAL (ANY CONSISTENT SET OF UNITS) RIN00590
C
C          DIMENSIONAL INPUTS (NOTE: U0=2.*NI*SI) RIN00600
C
C          NI          FREQUENCY (HZ) RIN00610
C          SI          STROKE (L) RIN00620
C          CI          CLEARANCE (L) RIN00630
C          ELI         BEARING LENGTH (L) RIN00640
C          ELII        LENGTH OF NON-BEARING PORTION OF RING (L) RIN00650
C          EI          LENGTH FROM END OF BEARING PRELOADED WITH POI (L) RIN00660
C          RI          RING RADIUS (L) RIN00670
C          TI          RING THICKNESS (L) RIN00680
C          CLI (D)     BEARING SLOPE RIN00690
C          POI         PRELOAD PRESSURE (F/L**2) RIN00700
C          PFI         RESERVOIR PRESSURE (F/L**2) RIN00710
C          RMUI        LUBRICANT VISCOSITY (FT/L**2) RIN00720
C          EMOD        RING MODULUS OF ELASTICITY (F/L**2) RIN00730
C          POIS        RING POISSON'S RATIO RIN00740
C
C          NON-DIMENSIONAL INPUTS RIN00750
C
C          AL          (TI*RI)**2/(12*ELI**4*(1.-POIS)**2)) RIN00760
C          BET         (T*RMUI*U0*RI**2*ELI)/(CI**3*TI*EMOD) RIN00770
C          EPS         LENGTH FROM END OF PRELOAD P0 (EI/ELI) RIN00780
C          ELI         LENGTH OF NON-BEARING PORTION OF RING (ELII/ELI) RIN00790
C          ELOS (D)    L/S LAND TO STROKE RATIO FOR STARVATION CALCULATION RIN00800
C          XLAM (D)    MULTIPLIES ELOS FOR INCREASED STARVATION RIN00810
C          C1 (D)     SLOPE OF BEARING RIN00820
C          P0         PRELOAD PRESSURE (CI**2/(6*RMUI*U0*ELI))*POI RIN00830
C          PF         RESERVOIR PRESSURE (CI**2/(6*RMUI*U0*ELI))*PFI RIN00840
C
C          NDX(3) (D) DELTA X INCREMENTS FOR RUK RIN00850
C          NDP(3) (D) " " " " " RIN00860
C
C          ICAV (D)   FLAG FOR CALCULATION OF CAVITATION (BACK STROKE) RIN00870
C                    0 - NO CAVITATION (NEG. PRESSURES ALLOWED) RIN00880
C                    1 - CAVITATION (NO NEG. PRESSURES) RIN00890
C                    2 - FIND ITS OWN SOLUTION RIN00900
C          IPR (D)   PRINT FLAG RIN00910
C                    0 - SHORT OUTPUT RIN00920
C                    1 - LONGER OUTPUT (PRINT PRESSURE PROFILE) RIN00930
C          NMAX (D)   MAX. ITERATIONS FOR *ZSCNT* RIN01000
C          NSIG (D)   NO. OF SIGNIFICANT DIGITS IN ACCURACY RIN01010
C                    OF SOLUTION USING *ZSCNT* RIN01020
C          XF(2)     INITIAL GUESS FOR H AND H' FOR FORWARD STROKE RIN01030
C          XB(2)     INITIAL GUESS FOR H AND H' FOR BACKWARD STROKE RIN01040
C
C          ----- RIN01060
C          ----- RIN01070
C          ----- RIN01080
C          IMPLICIT REAL*8 (A-H,O-Z) RIN01090
C

```

```

COMMON/BDIM /CI,ELI,CON1,IDIM RIN01110
COMMON/BPAR /AL,BET,PO,EPS,PF RIN01120
COMMON/BCD /XLOC(100),C(100),D(100),NDX(3),NDP(3),DX(3),DP(3) RIN01130
COMMON/BCOEFF/RK,RC,R1,R2,XCAV,ICAV RIN01140
COMMON/BINT /IELAS,IBACK,IPR RIN01150
COMMON/BLAST /X1LAST,X2LAST,F1LAST,F2LAST RIN01160
COMMON/BELAS /FORCE,HELAS,H1ELAS,H2ELAS,H3ELAS,W1(4),W2(4) RIN01170
COMMON/BPROF /C1 RIN01180
COMMON/BFLAG /IFLAG,IPLOAD RIN01190
C RIN01200
DIMENSION PAR(1),XX(2),XF(2),XB(2),WK(68) RIN01210
REAL*8 NFLOW,NI RIN01220
CHARACTER*60 STRING RIN01230
LOGICAL RECALC RIN01240
EXTERNAL EVAL,EVAL3,EVAL4 RIN01250
NAMELIST /INPUTS/ RIN01260
C RIN01270
C DIMENSIONAL INPUTS RIN01280
C RIN01290
+ STRING,NI,SI,C1I,CI,RMUI,ELI,POI,PFI,EI,ELI,RI,TI,EMOD,POIS RIN01300
C RIN01310
C NON-DIMENSIONAL INPUTS RIN01320
C RIN01330
+,BET,PO,EPS,AL,EL1,ELOS,XLAM, RIN01340
+NDX,NDP,PF,NMAX,IPR,NSIG,ICAV,XF,XB,C1,IDIM RIN01350
C RIN01360
PI=4.*DATAN(1.DO) RIN01370
C RIN01380
C DEFAULTS RIN01390
C RIN01400
IDIM=0 RIN01410
ELOS=0. RIN01420
XLAM=1. RIN01430
C1I=0. RIN01440
C1=0. RIN01450
NDX(1)=40 RIN01460
NDX(2)=25 RIN01470
NDX(3)=25 RIN01480
NDP(1)=10 RIN01490
NDP(2)=25 RIN01500
NDP(3)=25 RIN01510
NMAX=10 RIN01520
NSIG=4 RIN01530
ICAV=2 RIN01540
IPR=0 RIN01550
C RIN01560
IRUN=0 RIN01570
2001 READ(05,INPUTS,END=2000) RIN01580
C RIN01590
IRUN=IRUN+1 RIN01600
WRITE(6,('1* PUMPING RING ANALYSIS PROGRAM *',/)) RIN01610
WRITE(6,(' INPUTS')) RIN01620
WRITE(6,(1X,72('*'))') RIN01630
WRITE(6,(' STRING = ',A60)) STRING RIN01640
WRITE(6,(1X,72('*'))') RIN01650

```

```

IF(IDIM.NE.0)THEN
WRITE(6,(' DIMENSIONAL INPUTS',/))
WRITE(6,(' ELI =',E15.8,','; ELII =',E15.8,',';'))
+ELI,ELII
WRITE(6,(' EI =',E15.8,','; CI =',E15.8,',';'))EI,CIRIN01700
WRITE(6,(' CII =',E15.8,','; TI =',E15.8,',';'))
+CII,TI
WRITE(6,(' SI =',E15.8,','; RI =',E15.8,',';'))SI,RIRIN01730
WRITE(6,(' RMUI =',E15.8,','; NI =',E15.8,',';'))
+RMUI,NI
WRITE(6,(' '))
WRITE(6,(' POI =',E15.8,','; PFI =',E15.8,',';'))
+POI,PFI
WRITE(6,(' EMOD =',E15.8,','; POIS =',E15.8,',';'))
+EMOD,POIS
WRITE(6,(1X,72('*'))')
UO=2.*NI*SI
CON1=6.*RMUI*UO*ELI/CI**2
ELOS=ELI/SI
AL=(TI*(RI+TI*.5))**2/12./ELI**4/(1.-POIS**2)
BET=CON1*(RI+TI*.5)**2/CI/TI/EMOD
EPS=EI/ELI
EL1=ELII/ELI
C1=CII/CI*ELI
P0=POI/CON1
PF=PFI/CON1
END IF
IF(ELOS.GT.1.D-5)XLAM=.739/ELOS**.585
STARV=ELOS*XLAM
WRITE(6,(' IDIM =',I2,',';'))IDIM
WRITE(6,(' AL =',E15.8,','; BET =',E15.8,',';'))AL,BET
WRITE(6,(' EPS =',E15.8,','; P0 =',E15.8,',';'))EPS,P0
WRITE(6,(' EL1 =',E15.8,','; PF =',E15.8,',';'))EL1,PF
WRITE(6,(' STARV =',E15.8,','; C1 =',E15.8,',';'))STARV,C1
WRITE(6,(' '))
WRITE(6,(' NDX =',I5,','; NDX =',I5,','; NDX =',I5,',';'))NDX
WRITE(6,(' NDP =',I5,','; NDP =',I5,','; NDP =',I5,',';'))NDP
WRITE(6,(' '))
WRITE(6,(' NMAX =',I5,','; NSIG =',I5,',';'))NMAX,NSIG
WRITE(6,(' IPR =',I5,','; ICAV =',I5,',';'))IPR,ICAV
WRITE(6,(' XF =',2E15.5))XF(1),XF(2)
WRITE(6,(' XB =',2E15.5))XB(1),XB(2)
WRITE(6,(1X,72('*'))',/))
WRITE(6,(' OUTPUTS'))
WRITE(6,(1X,72('*'))')
C
IFLAG=0
IF(0..LT.EPS.AND.EPS.LT.1.)THEN
IFLAG=1
DX(1)=EL1/FLOAT(NDX(1))
DP(1)=EL1/FLOAT(NDP(1))
DX(2)=(1.-EPS)/FLOAT(NDX(2))
DP(2)=(1.-EPS)/FLOAT(NDP(2))
DX(3)=EPS/FLOAT(NDX(3))
DP(3)=EPS/FLOAT(NDP(3))

```

```
ELSE IF(EPS.EQ.1.)THEN
  IFLAG=2
  DX(1)=EL1/FLOAT(NDX(1))
  DP(1)=EL1/FLOAT(NDP(1))
  DX(2)=1./FLOAT(NDX(2))
  DP(2)=1./FLOAT(NDP(2))
  DX(3)=0.
  DP(3)=0.
ELSE IF(1..LT.EPS.AND.EPS.LT.EL1+1)THEN
  IFLAG=3
  DX(1)=(EL1+1.-EPS)/FLOAT(NDX(1))
  DP(1)=(EL1+1.-EPS)/FLOAT(NDP(1))
  DX(2)=(EPS-1.)/FLOAT(NDX(2))
  DP(2)=(EPS-1.)/FLOAT(NDP(2))
  DX(3)=1./FLOAT(NDX(3))
  DP(3)=1./FLOAT(NDP(3))
ELSE IF(EPS.EQ.1.+EL1)THEN
  IFLAG=4
  DX(1)=EL1/FLOAT(NDX(1))
  DP(1)=EL1/FLOAT(NDP(1))
  DX(2)=1./FLOAT(NDX(2))
  DP(2)=1./FLOAT(NDP(2))
  DX(3)=0.
  DP(3)=0.
END IF
IF(IFLAG.EQ.0)STOP
C
CALL CHECK(AL,BET,EPS,EL1,NDX,NDP,IRUN,RECALC)
IF(RECALC)THEN
  IF(IFLAG.LE.2)THEN
    CALL CALCD
  ELSE IF(IFLAG.EQ.3)THEN
    CALL CALCD3
  ELSE IF(IFLAG.EQ.4)THEN
    CALL CALCD4
  END IF
ELSE
C   WRITE(6,'(20X,' * NOTE: C,D NOT RECALCULATED ! *')')
END IF
C
C   ELASTIC ANALYSIS
C
WRITE(6,'('' ',72(''-'))')
WRITE(6,'(26X,' * ELASTIC SOLUTION *',/))
CALL ELAS
C
C   STEADY STATE ANALYSIS
C
IELAS=0
C
C   FORWARD FLOW SOLUTION
C
IBACK=0
WRITE(6,'('' ',72(''-'))')
WRITE(6,'(23X,' * PUMPING FLOW SOLUTION *',/))
```

RIN02210
RIN02220
RIN02230
RIN02240
RIN02250
RIN02260
RIN02270
RIN02280
RIN02290
RIN02300
RIN02310
RIN02320
RIN02330
RIN02340
RIN02350
RIN02360
RIN02370
RIN02380
RIN02390
RIN02400
RIN02410
RIN02420
RIN02430
RIN02440
RIN02450
RIN02460
RIN02470
RIN02480
RIN02490
RIN02500
RIN02510
RIN02520
RIN02530
RIN02540
RIN02550
RIN02560
RIN02570
RIN02580
RIN02590
RIN02600
RIN02610
RIN02620
RIN02630
RIN02640
RIN02650
RIN02660
RIN02670
RIN02680
RIN02690
RIN02700
RIN02710
RIN02720
RIN02730
RIN02740
RIN02750

```

IF(XF(1)**2+XF(2)**2.LT.1.D-5)THEN
  XX(1)=1.-HELAS+.01
  XX(2)=C1-H1ELAS+.01
ELSE
  XX(1)=1.-XF(1)
  XX(2)=C1-XF(2)
END IF
IF(IFLAG.LE.2)THEN
  CALL ZSCNT(EVAL,NSIG,2,NMAX,PAR,XX,FNORM,WK,IER)
ELSE IF(IFLAG.EQ.3)THEN
  CALL ZSCNT(EVAL3,NSIG,2,NMAX,PAR,XX,FNORM,WK,IER)
ELSE IF(IFLAG.EQ.4)THEN
  CALL ZSCNT(EVAL4,NSIG,2,NMAX,PAR,XX,FNORM,WK,IER)
END IF
IF(IER.NE.0)CALL ERRMSG
CALL PRTOU(XX)
XF(1)=1.-XX(1)
XF(2)=C1-XX(2)
FFLOW=RK
C
C
C
BACK FLOW SOLUTION
IBACK=1
WRITE(6,('( ' ',72(' '-'))')
WRITE(6,(25X,' * BACK FLOW SOLUTION *',/))
IF(XB(1)**2+XB(2)**2.LT.1.D-5)THEN
  XX(1)=1.-HELAS
  XX(2)=C1-H1ELAS
ELSE
  XX(1)=1.-XB(1)
  XX(2)=C1-XB(2)
END IF
BFLOW=0.
IF(FORCE.LT.1.D-8)THEN
  IF(IFLAG.LE.2)THEN
    CALL ZSCNT(EVAL,NSIG,2,NMAX,PAR,XX,FNORM,WK,IER)
  ELSE IF(IFLAG.EQ.3)THEN
    CALL ZSCNT(EVAL3,NSIG,2,NMAX,PAR,XX,FNORM,WK,IER)
  ELSE IF(IFLAG.EQ.4)THEN
    CALL ZSCNT(EVAL4,NSIG,2,NMAX,PAR,XX,FNORM,WK,IER)
  END IF
  IF(IER.NE.0)CALL ERRMSG
  BFLOW=RK
END IF
CALL PRTOU(XX)
XB(1)=1.-XX(1)
XB(2)=C1-XX(2)
C
3000 WRITE(6,('( ' ',72(' '-'))')
CONTINUE
NFLOW=FFLOW-BFLOW
WRITE(6,(32X,' * FLOWS *',/))
WRITE(6,(5X,' PUMPING',10X,' BACK',11X,' XCAV',9X
+ , ' NET FLOW',3X,' STARVED NET FLOW'))
SFLOW=NFLOW

```

RIN02760
RIN02770
RIN02780
RIN02790
RIN02800
RIN02810
RIN02820
RIN02830
RIN02840
RIN02850
RIN02860
RIN02870
RIN02880
RIN02890
RIN02900
RIN02910
RIN02920
RIN02930
RIN02940
RIN02950
RIN02960
RIN02970
RIN02980
RIN02990
RIN03000
RIN03010
RIN03020
RIN03030
RIN03040
RIN03050
RIN03060
RIN03070
RIN03080
RIN03090
RIN03100
RIN03110
RIN03120
RIN03130
RIN03140
RIN03150
RIN03160
RIN03170
RIN03180
RIN03190
RIN03200
RIN03210
RIN03220
RIN03230
RIN03240
RIN03250
RIN03260
RIN03270
RIN03280
RIN03290
RIN03300

```

HSTV=XB(1) RIN03310
DHSTV=XB(2) RIN03320
  IF(IBACK.EQ.1.AND.FORCE.GT.1.D-8)THEN RIN03330
    XCAV=1. RIN03340
    HSTV=HELAS RIN03350
    DHSTV=H1ELAS RIN03360
  END IF RIN03370
  IF(XCAV.GT.1.D-6.AND.XCAV.LT.1.000001)SFLOW=FFLOW*(1.+ RIN03380
+2.*ELOS*XLAM*XCAV**2*DHSTV/(2.*HSTV-DHSTV*(2.-XCAV)))-BFLOW RIN03390
  WRITE(6,'(5E15.6)')FFLOW,BFLOW RIN03400
+ ,XCAV,NFLOW,SFLOW RIN03410
  WRITE(6,'(1X,72(''*''),/)'') RIN03420
  GOTO 2001 RIN03430
2000 CALL EXIT RIN03440
  END RIN03450

SUBROUTINE AIN(A,B) RIN03460
C----- RIN03470
C RIN03480
C RIN03490
C FUNCTION - CALCULATE INVERSE OF 2X2 MATRIX RIN03500
C RIN03510
C RESTRICTIONS - RIN03520
C RIN03530
C REMARKS - RIN03540
C RIN03550
C EXTERNAL REFERENCES - NONE RIN03560
C RIN03570
C ARGUMENT DEFINITION: RIN03580
C NAME DESCRIPTION RIN03590
C A INPUT MATRIX RIN03600
C B OUTPUT MATRIX (INVERSE OF A) RIN03610
C RIN03620
C RIN03630
C----- RIN03640
IMPLICIT REAL*8 (A-H,O-Z) RIN03650
DIMENSION A(2,2),B(2,2) RIN03660
D=A(1,1)*A(2,2)-A(2,1)*A(1,2) RIN03670
B(1,1)=A(2,2)/D RIN03680
B(2,2)=A(1,1)/D RIN03690
B(1,2)=-A(1,2)/D RIN03700
B(2,1)=-A(2,1)/D RIN03710
RETURN RIN03720
END RIN03730

```

```

SUBROUTINE AMU(A,B,C)
C-----RIN03740
C-----RIN03750
C-----RIN03760
C-----RIN03770
C     FUNCTION          - PERFORM MATRIX MULTIPLICATION OF 2X2 MATRIX RIN03780
C-----RIN03790
C     RESTRICTIONS      - RIN03800
C-----RIN03810
C     REMARKS           - RIN03820
C-----RIN03830
C     EXTERNAL REFERENCES - NONE RIN03840
C-----RIN03850
C     ARGUMENT DEFINITION: RIN03860
C     NAME              DESCRIPTION RIN03870
C     A                 INPUT MATRIX RIN03880
C     B                 INPUT MATRIX RIN03890
C     C                 OUTPUT MATRIX (C=AXB) RIN03900
C-----RIN03910
C-----RIN03920
C-----RIN03930
IMPLICIT REAL*8 (A-H,O-Z) RIN03940
DIMENSION A(2,2),B(2,2),C(2,2) RIN03950
C(1,1)=A(1,1)*B(1,1)+A(1,2)*B(2,1) RIN03960
C(2,1)=A(2,1)*B(1,1)+A(2,2)*B(2,1) RIN03970
C(1,2)=A(1,1)*B(1,2)+A(1,2)*B(2,2) RIN03980
C(2,2)=A(2,1)*B(1,2)+A(2,2)*B(2,2) RIN03990
RETURN RIN04000
END RIN04010
```

```

SUBROUTINE CALCD
C>-----RIN04020
C-----RIN04030
C-----RIN04040
C-----RIN04050
C    FUNCTION          - CALCULATE INFLUENCE COEFFICIENTS C,D RIN04060
C                      FOR H AND H' AT PSI=1 RIN04070
C-----RIN04080
C    RESTRICTIONS      - FOR MULTIPLE RUNS, ONLY COMPUTED WHEN RIN04090
C                      CERTAIN PARAMETERS CHANGE. ALWAYS RIN04100
C                      COMPUTED AT LEAST ONCE. RIN04110
C-----RIN04120
C    REMARKS           - NOTE VARIABLES PASSED IN COMMON RIN04130
C-----RIN04140
C    EXTERNAL REFERENCES - AIN,AMU,RUK RIN04150
C-----RIN04160
C-----RIN04170
C>-----RIN04180
IMPLICIT REAL*8 (A-H,O-Z) RIN04190
COMMON/BCD /XLOC(100),C(100),D(100),NDX(3),NDP(3),DX(3),DP(3) RIN04200
COMMON/BELAS /FORCE,HELAS,H1ELAS,H2ELAS,H3ELAS,W1(4),W2(4) RIN04210
COMMON/BPAR /AL,BET,PO,EPS,PF,S,U,DT RIN04220
COMMON/BFLAG /IFLAG,IPLoad RIN04230
DIMENSION YO(4),YT(4),DJ(4),CKJ(4,4) RIN04240
DIMENSION XLEN(3) RIN04250
DIMENSION A1(2,2,100),A2(2,2,100) RIN04260
DIMENSION B1(2,2,100),B2(2,2,100) RIN04270
DIMENSION A1I(2,2),AT(2,2) RIN04280
EXTERNAL DFN1 RIN04290
C-----RIN04300
XLEN(1)=DX(1)*NDX(1) RIN04310
XLEN(2)=DX(2)*NDX(2) RIN04320
XLEN(3)=DX(3)*NDX(3) RIN04330
EL1=XLEN(1) RIN04340
YO(1)=0. RIN04350
YO(2)=0. RIN04360
YO(3)=1. RIN04370
YO(4)=0. RIN04380
CALL RUK(DX(1),XLEN(1),-EL1,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN04390
CALL RUK(DX(2)/2.,DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN04400
XLOC(1)=XNN RIN04410
A1(1,1,1)=YO(1) RIN04420
A1(2,1,1)=YO(2) RIN04430
A2(1,1,1)=YO(3) RIN04440
A2(2,1,1)=YO(4) RIN04450
DO 100 I=2,NDP(2) RIN04460
CALL RUK(DX(2),DP(2),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN04470
XLOC(I)=XNN RIN04480
A1(1,1,I)=YO(1) RIN04490
A1(2,1,I)=YO(2) RIN04500
A2(1,1,I)=YO(3) RIN04510
100 A2(2,1,I)=YO(4) RIN04520
CALL RUK(DX(2)/2.,DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN04530
IF(IFLAG.EQ.2)GOTO 111 RIN04540
CALL RUK(DX(3)/2.,DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN04550
XLOC(NDP(2)+1)=XNN RIN04560

```

```
A1(1,1,NDP(2)+1)=YO(1) RIN04570
A1(2,1,NDP(2)+1)=YO(2) RIN04580
A2(1,1,NDP(2)+1)=YO(3) RIN04590
A2(2,1,NDP(2)+1)=YO(4) RIN04600
DO 101 I=2,NDP(3) RIN04610
CALL RUK(DX(3),DP(3),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN04620
XLOC(NDP(2)+I)=XNN RIN04630
A1(1,1,NDP(2)+I)=YO(1) RIN04640
A1(2,1,NDP(2)+I)=YO(2) RIN04650
A2(1,1,NDP(2)+I)=YO(3) RIN04660
101 A2(2,1,NDP(2)+I)=YO(4) RIN04670
CALL RUK(DX(3)/2.,DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN04680
111 W1(1)=YO(1) RIN04690
W1(2)=YO(2) RIN04700
W1(3)=YO(3) RIN04710
W1(4)=YO(4) RIN04720
C RIN04730
YO(1)=0. RIN04740
YO(2)=0. RIN04750
YO(3)=0. RIN04760
YO(4)=1. RIN04770
CALL RUK(DX(1),XLEN(1),-EL1,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN04780
CALL RUK(DX(2)/2.,DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN04790
A1(1,2,1)=YO(1) RIN04800
A1(2,2,1)=YO(2) RIN04810
A2(1,2,1)=YO(3) RIN04820
A2(2,2,1)=YO(4) RIN04830
DO 200 I=2,NDP(2) RIN04840
CALL RUK(DX(2),DP(2),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN04850
A1(1,2,I)=YO(1) RIN04860
A1(2,2,I)=YO(2) RIN04870
A2(1,2,I)=YO(3) RIN04880
200 A2(2,2,I)=YO(4) RIN04890
CALL RUK(DX(2)/2.,DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN04900
IF(IFLAG.EQ.2)GOTO 112 RIN04910
CALL RUK(DX(3)/2.,DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN04920
A1(1,2,NDP(2)+1)=YO(1) RIN04930
A1(2,2,NDP(2)+1)=YO(2) RIN04940
A2(1,2,NDP(2)+1)=YO(3) RIN04950
A2(2,2,NDP(2)+1)=YO(4) RIN04960
DO 201 I=2,NDP(3) RIN04970
CALL RUK(DX(3),DP(3),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN04980
A1(1,2,NDP(2)+I)=YO(1) RIN04990
A1(2,2,NDP(2)+I)=YO(2) RIN05000
A2(1,2,NDP(2)+I)=YO(3) RIN05010
201 A2(2,2,NDP(2)+I)=YO(4) RIN05020
CALL RUK(DX(3)/2.,DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN05030
112 W2(1)=YO(1) RIN05040
W2(2)=YO(2) RIN05050
W2(3)=YO(3) RIN05060
W2(4)=YO(4) RIN05070
C RIN05080
YO(1)=1. RIN05090
YO(2)=0. RIN05100
YO(3)=0. RIN05110
```

```
YO(4)=0. RIN05120
XNN=1. RIN05130
IF(IFLAG.EQ.2)GOTO 113 RIN05140
CALL RUK(-DX(3)/2.,-DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN05150
B1(1,1,NDP(2)+NDP(3))=YO(1) RIN05160
B1(2,1,NDP(2)+NDP(3))=YO(2) RIN05170
B2(1,1,NDP(2)+NDP(3))=YO(3) RIN05180
B2(2,1,NDP(2)+NDP(3))=YO(4) RIN05190
DO 300 II=2,NDP(3) RIN05200
I=NDP(2)+NDP(3)-II+1 RIN05210
CALL RUK(-DX(3),-DP(3),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN05220
B1(1,1,I)=YO(1) RIN05230
B1(2,1,I)=YO(2) RIN05240
B2(1,1,I)=YO(3) RIN05250
300 B2(2,1,I)=YO(4) RIN05260
CALL RUK(-DX(3)/2.,-DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN05270
113 CALL RUK(-DX(2)/2.,-DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN05280
B1(1,1,NDP(2))=YO(1) RIN05290
B1(2,1,NDP(2))=YO(2) RIN05300
B2(1,1,NDP(2))=YO(3) RIN05310
B2(2,1,NDP(2))=YO(4) RIN05320
DO 301 II=2,NDP(2) RIN05330
I=NDP(2)-II+1 RIN05340
CALL RUK(-DX(2),-DP(2),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN05350
B1(1,1,I)=YO(1) RIN05360
B1(2,1,I)=YO(2) RIN05370
B2(1,1,I)=YO(3) RIN05380
301 B2(2,1,I)=YO(4) RIN05390
C RIN05400
YO(1)=0. RIN05410
YO(2)=1. RIN05420
YO(3)=0. RIN05430
YO(4)=0. RIN05440
XNN=1. RIN05450
IF(IFLAG.EQ.2)GOTO 114 RIN05460
CALL RUK(-DX(3)/2.,-DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN05470
B1(1,2,NDP(2)+NDP(3))=YO(1) RIN05480
B1(2,2,NDP(2)+NDP(3))=YO(2) RIN05490
B2(1,2,NDP(2)+NDP(3))=YO(3) RIN05500
B2(2,2,NDP(2)+NDP(3))=YO(4) RIN05510
DO 400 II=2,NDP(3) RIN05520
I=NDP(2)+NDP(3)-II+1 RIN05530
CALL RUK(-DX(3),-DP(3),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN05540
B1(1,2,I)=YO(1) RIN05550
B1(2,2,I)=YO(2) RIN05560
B2(1,2,I)=YO(3) RIN05570
400 B2(2,2,I)=YO(4) RIN05580
CALL RUK(-DX(3)/2.,-DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN05590
114 CALL RUK(-DX(2)/2.,-DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN05600
B1(1,2,NDP(2))=YO(1) RIN05610
B1(2,2,NDP(2))=YO(2) RIN05620
B2(1,2,NDP(2))=YO(3) RIN05630
B2(2,2,NDP(2))=YO(4) RIN05640
DO 401 II=2,NDP(2) RIN05650
I=NDP(2)-II+1 RIN05660
```

	CALL RUK(-DX(2),-DP(2),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)	RIN05670
	B1(1,2,I)=YO(1)	RIN05680
	B1(2,2,I)=YO(2)	RIN05690
	B2(1,2,I)=YO(3)	RIN05700
401	B2(2,2,I)=YO(4)	RIN05710
C		RIN05720
	DO 500 I=1,NDP(2)	RIN05730
	CALL AIN(A1(1,1,I),A1I)	RIN05740
	CALL AMU(A2(1,1,I),A1I,AT)	RIN05750
	CALL AMU(AT,B1(1,1,I),A1(1,1,I))	RIN05760
	DO 21 II=1,2	RIN05770
	DO 21 JJ=1,2	RIN05780
21	AT(II,JJ)=B2(II,JJ,I)-A1(II,JJ,I)	RIN05790
	CALL AIN(AT,B2(1,1,I))	RIN05800
	C(I)=B2(1,2,I)/AL*DP(2)	RIN05810
	D(I)=B2(2,2,I)/AL*DP(2)	RIN05820
500	CONTINUE	RIN05830
	IF(IFLAG.EQ.2)RETURN	RIN05840
	INDEX1=NDP(2)+1	RIN05850
	INDEX2=NDP(2)+NDP(3)	RIN05860
	DO 501 I=INDEX1,INDEX2	RIN05870
	CALL AIN(A1(1,1,I),A1I)	RIN05880
	CALL AMU(A2(1,1,I),A1I,AT)	RIN05890
	CALL AMU(AT,B1(1,1,I),A1(1,1,I))	RIN05900
	DO 22 II=1,2	RIN05910
	DO 22 JJ=1,2	RIN05920
22	AT(II,JJ)=B2(II,JJ,I)-A1(II,JJ,I)	RIN05930
	CALL AIN(AT,B2(1,1,I))	RIN05940
	C(I)=B2(1,2,I)/AL*DP(3)	RIN05950
	D(I)=B2(2,2,I)/AL*DP(3)	RIN05960
501	CONTINUE	RIN05970
C		RIN05980
	RETURN	RIN05990
	END	RIN06000

```

SUBROUTINE CALCD3
C>-----RIN06010
C RIN06020
C RIN06030
C RIN06040
C FUNCTION - CALCULATE INFLUENCE COEFFICIENTS C,D RIN06050
C FOR H AND H' AT PSI=1 RIN06060
C RIN06070
C RESTRICTIONS - FOR MULTIPLE RUNS, ONLY COMPUTED WHEN RIN06080
C CERTAIN PARAMETERS CHANGE. ALWAYS RIN06090
C COMPUTED AT LEAST ONCE. RIN06100
C RIN06110
C REMARKS - NOTE VARIABLES PASSED IN COMMON RIN06120
C RIN06130
C EXTERNAL REFERENCES - AIN,AMU, RUK RIN06140
C RIN06150
C RIN06160
C>-----RIN06170
IMPLICIT REAL*8 (A-H,O-Z) RIN06180
COMMON/BCD /XLOC(100),C(100),D(100),NDX(3),NDP(3),DX(3),DP(3) RIN06190
COMMON/BELAS /FORCE,HELAS,H1ELAS,H2ELAS,H3ELAS,W1(4),W2(4) RIN06200
COMMON/BPAR /AL,BET,PO,EPS,PF,S,U,DT RIN06210
COMMON/BFLAG /IFLAG,IPLOAD RIN06220
DIMENSION YO(4),YT(4),DJ(4),CKJ(4,4) RIN06230
DIMENSION XLEN(3) RIN06240
DIMENSION A1(2,2,100),A2(2,2,100) RIN06250
DIMENSION B1(2,2,100),B2(2,2,100) RIN06260
DIMENSION A1I(2,2),AT(2,2) RIN06270
EXTERNAL DFN1 RIN06280
C RIN06290
XLEN(1)=DX(1)*NDX(1) RIN06300
XLEN(2)=DX(2)*NDX(2) RIN06310
XLEN(3)=DX(3)*NDX(3) RIN06320
EL1=XLEN(1)+XLEN(2) RIN06330
YO(1)=0. RIN06340
YO(2)=0. RIN06350
YO(3)=1. RIN06360
YO(4)=0. RIN06370
CALL RUK(DX(1),XLEN(1),-EL1,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN06380
CALL RUK(DX(2)/2.,DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN06390
XLOC(1)=XNN RIN06400
A1(1,1,1)=YO(1) RIN06410
A1(2,1,1)=YO(2) RIN06420
A2(1,1,1)=YO(3) RIN06430
A2(2,1,1)=YO(4) RIN06440
DO 100 I=2,NDP(2) RIN06450
CALL RUK(DX(2),DP(2),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN06460
XLOC(I)=XNN RIN06470
A1(1,1,I)=YO(1) RIN06480
A1(2,1,I)=YO(2) RIN06490
A2(1,1,I)=YO(3) RIN06500
100 A2(2,1,I)=YO(4) RIN06510
CALL RUK(DX(2)/2.,DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN06520
CALL RUK(DX(3)/2.,DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN06530
XLOC(NDP(2)+1)=XNN RIN06540
A1(1,1,NDP(2)+1)=YO(1) RIN06550

```

```
A1(2,1,NDP(2)+1)=YO(2)
A2(1,1,NDP(2)+1)=YO(3)
A2(2,1,NDP(2)+1)=YO(4)
DO 101 I=2,NDP(3)
CALL RUK(DX(3),DP(3),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
XLOC(NDP(2)+I)=XNN
A1(1,1,NDP(2)+I)=YO(1)
A1(2,1,NDP(2)+I)=YO(2)
A2(1,1,NDP(2)+I)=YO(3)
101 A2(2,1,NDP(2)+I)=YO(4)
CALL RUK(DX(3)/2.,DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
W1(1)=YO(1)
W1(2)=YO(2)
W1(3)=YO(3)
W1(4)=YO(4)
C
YO(1)=0.
YO(2)=0.
YO(3)=0.
YO(4)=1.
CALL RUK(DX(1),XLEN(1),-EL1,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
CALL RUK(DX(2)/2.,DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
A1(1,2,1)=YO(1)
A1(2,2,1)=YO(2)
A2(1,2,1)=YO(3)
A2(2,2,1)=YO(4)
DO 200 I=2,NDP(2)
CALL RUK(DX(2),DP(2),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
A1(1,2,I)=YO(1)
A1(2,2,I)=YO(2)
A2(1,2,I)=YO(3)
200 A2(2,2,I)=YO(4)
CALL RUK(DX(2)/2.,DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
CALL RUK(DX(3)/2.,DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
A1(1,2,NDP(2)+1)=YO(1)
A1(2,2,NDP(2)+1)=YO(2)
A2(1,2,NDP(2)+1)=YO(3)
A2(2,2,NDP(2)+1)=YO(4)
DO 201 I=2,NDP(3)
CALL RUK(DX(3),DP(3),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
A1(1,2,NDP(2)+I)=YO(1)
A1(2,2,NDP(2)+I)=YO(2)
A2(1,2,NDP(2)+I)=YO(3)
201 A2(2,2,NDP(2)+I)=YO(4)
CALL RUK(DX(3)/2.,DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
W2(1)=YO(1)
W2(2)=YO(2)
W2(3)=YO(3)
W2(4)=YO(4)
C
YO(1)=1.
YO(2)=0.
YO(3)=0.
YO(4)=0.
XNN=1.
```

```
RIN06560
RIN06570
RIN06580
RIN06590
RIN06600
RIN06610
RIN06620
RIN06630
RIN06640
RIN06650
RIN06660
RIN06670
RIN06680
RIN06690
RIN06700
RIN06710
RIN06720
RIN06730
RIN06740
RIN06750
RIN06760
RIN06770
RIN06780
RIN06790
RIN06800
RIN06810
RIN06820
RIN06830
RIN06840
RIN06850
RIN06860
RIN06870
RIN06880
RIN06890
RIN06900
RIN06910
RIN06920
RIN06930
RIN06940
RIN06950
RIN06960
RIN06970
RIN06980
RIN06990
RIN07000
RIN07010
RIN07020
RIN07030
RIN07040
RIN07050
RIN07060
RIN07070
RIN07080
RIN07090
RIN07100
```

```
CALL RUK(-DX(3)/2.,-DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN07110
B1(1,1,NDP(2)+NDP(3))=YO(1) RIN07120
B1(2,1,NDP(2)+NDP(3))=YO(2) RIN07130
B2(1,1,NDP(2)+NDP(3))=YO(3) RIN07140
B2(2,1,NDP(2)+NDP(3))=YO(4) RIN07150
DO 300 II=2,NDP(3) RIN07160
I=NDP(2)+NDP(3)-II+1 RIN07170
CALL RUK(-DX(3),-DP(3),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN07180
B1(1,1,I)=YO(1) RIN07190
B1(2,1,I)=YO(2) RIN07200
B2(1,1,I)=YO(3) RIN07210
300 B2(2,1,I)=YO(4) RIN07220
CALL RUK(-DX(3)/2.,-DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN07230
CALL RUK(-DX(2)/2.,-DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN07240
B1(1,1,NDP(2))=YO(1) RIN07250
B1(2,1,NDP(2))=YO(2) RIN07260
B2(1,1,NDP(2))=YO(3) RIN07270
B2(2,1,NDP(2))=YO(4) RIN07280
DO 301 II=2,NDP(2) RIN07290
I=NDP(2)-II+1 RIN07300
CALL RUK(-DX(2),-DP(2),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN07310
B1(1,1,I)=YO(1) RIN07320
B1(2,1,I)=YO(2) RIN07330
B2(1,1,I)=YO(3) RIN07340
301 B2(2,1,I)=YO(4) RIN07350
C RIN07360
YO(1)=0. RIN07370
YO(2)=1. RIN07380
YO(3)=0. RIN07390
YO(4)=0. RIN07400
XNN=1. RIN07410
CALL RUK(-DX(3)/2.,-DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN07420
B1(1,2,NDP(2)+NDP(3))=YO(1) RIN07430
B1(2,2,NDP(2)+NDP(3))=YO(2) RIN07440
B2(1,2,NDP(2)+NDP(3))=YO(3) RIN07450
B2(2,2,NDP(2)+NDP(3))=YO(4) RIN07460
DO 400 II=2,NDP(3) RIN07470
I=NDP(2)+NDP(3)-II+1 RIN07480
CALL RUK(-DX(3),-DP(3),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN07490
B1(1,2,I)=YO(1) RIN07500
B1(2,2,I)=YO(2) RIN07510
B2(1,2,I)=YO(3) RIN07520
400 B2(2,2,I)=YO(4) RIN07530
CALL RUK(-DX(3)/2.,-DP(3)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN07540
114 CALL RUK(-DX(2)/2.,-DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN07550
B1(1,2,NDP(2))=YO(1) RIN07560
B1(2,2,NDP(2))=YO(2) RIN07570
B2(1,2,NDP(2))=YO(3) RIN07580
B2(2,2,NDP(2))=YO(4) RIN07590
DO 401 II=2,NDP(2) RIN07600
I=NDP(2)-II+1 RIN07610
CALL RUK(-DX(2),-DP(2),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN07620
B1(1,2,I)=YO(1) RIN07630
B1(2,2,I)=YO(2) RIN07640
B2(1,2,I)=YO(3) RIN07650
```

401	B2(2,2,I)=YO(4)	RIN07660
C		RIN07670
	DO 500 I=1,NDP(2)	RIN07680
	CALL AIN(A1(1,1,I),AII)	RIN07690
	CALL AMU(A2(1,1,I),AII,AT)	RIN07700
	CALL AMU(AT,B1(1,1,I),A1(1,1,I))	RIN07710
	DO 21 II=1,2	RIN07720
	DO 21 JJ=1,2	RIN07730
21	AT(II,JJ)=B2(II,JJ,I)-A1(II,JJ,I)	RIN07740
	CALL AIN(AT,B2(1,1,I))	RIN07750
	C(I)=B2(1,2,I)/AL*DP(2)	RIN07760
	D(I)=B2(2,2,I)/AL*DP(2)	RIN07770
500	CONTINUE	RIN07780
	INDEX1=NDP(2)+1	RIN07790
	INDEX2=NDP(2)+NDP(3)	RIN07800
	DO 501 I=INDEX1,INDEX2	RIN07810
	CALL AIN(A1(1,1,I),AII)	RIN07820
	CALL AMU(A2(1,1,I),AII,AT)	RIN07830
	CALL AMU(AT,B1(1,1,I),A1(1,1,I))	RIN07840
	DO 22 II=1,2	RIN07850
	DO 22 JJ=1,2	RIN07860
22	AT(II,JJ)=B2(II,JJ,I)-A1(II,JJ,I)	RIN07870
	CALL AIN(AT,B2(1,1,I))	RIN07880
	C(I)=B2(1,2,I)/AL*DP(3)	RIN07890
	D(I)=B2(2,2,I)/AL*DP(3)	RIN07900
501	CONTINUE	RIN07910
C		RIN07920
	RETURN	RIN07930
	END	RIN07940

```

SUBROUTINE CALCD4
-----
C>-----RIN07950
C-----RIN07960
C-----RIN07970
C-----RIN07980
C    FUNCTION          - CALCULATE INFLUENCE COEFFICIENTS C,D    RIN07990
C                      FOR H AND H' AT PSI=1                    RIN08000
C-----RIN08010
C    RESTRICTIONS      - FOR MULTIPLE RUNS, ONLY COMPUTED WHEN  RIN08020
C                      CERTAIN PARAMETERS CHANGE. ALWAYS        RIN08030
C                      COMPUTED AT LEAST ONCE.                  RIN08040
C-----RIN08050
C    REMARKS           - NOTE VARIABLES PASSED IN COMMON        RIN08060
C-----RIN08070
C    EXTERNAL REFERENCES - AIN,AMU,RUK                          RIN08080
C-----RIN08090
C-----RIN08100
C>-----RIN08110
    IMPLICIT REAL*8 (A-H,O-Z)                                   RIN08120
    COMMON/BCD /XLOC(100),C(100),D(100),NDX(3),NDP(3),DX(3),DP(3) RIN08130
    COMMON/BELAS /FORCE,HELAS,H1ELAS,H2ELAS,H3ELAS,W1(4),W2(4)  RIN08140
    COMMON/BPAR /AL,BET,PO,EPS,PF,S,U,DT                       RIN08150
    COMMON/BFLAG /IFLAG,IPLOAD                                  RIN08160
    DIMENSION YO(4),YT(4),DJ(4),CKJ(4,4)                       RIN08170
    DIMENSION XLEN(3)                                           RIN08180
    DIMENSION A1(2,2,100),A2(2,2,100)                          RIN08190
    DIMENSION B1(2,2,100),B2(2,2,100)                          RIN08200
    DIMENSION A1I(2,2),AT(2,2)                                  RIN08210
    EXTERNAL DFN1                                               RIN08220
C-----RIN08230
C    XLEN(1)=DX(1)*NDX(1)                                       RIN08240
C    XLEN(2)=DX(2)*NDX(2)                                       RIN08250
C    XLEN(3)=DX(3)*NDX(3)                                       RIN08260
C    EL1=XLEN(1)                                                 RIN08270
C    YO(1)=0.                                                    RIN08280
C    YO(2)=0.                                                    RIN08290
C    YO(3)=1.                                                    RIN08300
C    YO(4)=0.                                                    RIN08310
C    CALL RUK(DX(1)/2.,DP(1)/2.,-EL1,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN08320
C    XLOC(1)=XNN                                                 RIN08330
C    A1(1,1,1)=YO(1)                                             RIN08340
C    A1(2,1,1)=YO(2)                                             RIN08350
C    A2(1,1,1)=YO(3)                                             RIN08360
C    A2(2,1,1)=YO(4)                                             RIN08370
C    DO 100 I=2,NDP(1)                                           RIN08380
C    CALL RUK(DX(1),DP(1),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)    RIN08390
C    XLOC(I)=XNN                                                 RIN08400
C    A1(1,1,I)=YO(1)                                             RIN08410
C    A1(2,1,I)=YO(2)                                             RIN08420
C    A2(1,1,I)=YO(3)                                             RIN08430
100  A2(2,1,I)=YO(4)                                             RIN08440
C    CALL RUK(DX(1)/2.,DP(1)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN08450
C    CALL RUK(DX(2)/2.,DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN08460
C    XLOC(NDP(1)+1)=XNN                                          RIN08470
C    A1(1,1,NDP(1)+1)=YO(1)                                       RIN08480
C    A1(2,1,NDP(1)+1)=YO(2)                                       RIN08490

```

```

A2(1,1,NDP(1)+1)=YO(3)
A2(2,1,NDP(1)+1)=YO(4)
DO 101 I=2,NDP(2)
CALL RUK(DX(2),DP(2),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
XLOC(NDP(1)+I)=XNN
A1(1,1,NDP(1)+I)=YO(1)
A1(2,1,NDP(1)+I)=YO(2)
A2(1,1,NDP(1)+I)=YO(3)
101 A2(2,1,NDP(1)+I)=YO(4)
CALL RUK(DX(2)/2.,DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
W1(1)=YO(1)
W1(2)=YO(2)
W1(3)=YO(3)
W1(4)=YO(4)
C
YO(1)=0.
YO(2)=0.
YO(3)=0.
YO(4)=1.
CALL RUK(DX(1)/2.,DP(1)/2.,-EL1,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
A1(1,2,1)=YO(1)
A1(2,2,1)=YO(2)
A2(1,2,1)=YO(3)
A2(2,2,1)=YO(4)
DO 200 I=2,NDP(1)
CALL RUK(DX(1),DP(1),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
A1(1,2,I)=YO(1)
A1(2,2,I)=YO(2)
A2(1,2,I)=YO(3)
200 A2(2,2,I)=YO(4)
CALL RUK(DX(1)/2.,DP(1)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
CALL RUK(DX(2)/2.,DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
A1(1,2,NDP(1)+1)=YO(1)
A1(2,2,NDP(1)+1)=YO(2)
A2(1,2,NDP(1)+1)=YO(3)
A2(2,2,NDP(1)+1)=YO(4)
DO 201 I=2,NDP(2)
CALL RUK(DX(2),DP(2),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
A1(1,2,NDP(1)+I)=YO(1)
A1(2,2,NDP(1)+I)=YO(2)
A2(1,2,NDP(1)+I)=YO(3)
201 A2(2,2,NDP(1)+I)=YO(4)
CALL RUK(DX(2)/2.,DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
W2(1)=YO(1)
W2(2)=YO(2)
W2(3)=YO(3)
W2(4)=YO(4)
C
YO(1)=1.
YO(2)=0.
YO(3)=0.
YO(4)=0.
XNN=1.
CALL RUK(-DX(2)/2.,-DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4)
B1(1,1,NDP(1)+NDP(2))=YO(1)

```

```

RIN08500
RIN08510
RIN08520
RIN08530
RIN08540
RIN08550
RIN08560
RIN08570
RIN08580
RIN08590
RIN08600
RIN08610
RIN08620
RIN08630
RIN08640
RIN08650
RIN08660
RIN08670
RIN08680
RIN08690
RIN08700
RIN08710
RIN08720
RIN08730
RIN08740
RIN08750
RIN08760
RIN08770
RIN08780
RIN08790
RIN08800
RIN08810
RIN08820
RIN08830
RIN08840
RIN08850
RIN08860
RIN08870
RIN08880
RIN08890
RIN08900
RIN08910
RIN08920
RIN08930
RIN08940
RIN08950
RIN08960
RIN08970
RIN08980
RIN08990
RIN09000
RIN09010
RIN09020
RIN09030
RIN09040

```

```
B1(2,1,NDP(1)+NDP(2))=YO(2)          RIN09050
B2(1,1,NDP(1)+NDP(2))=YO(3)          RIN09060
B2(2,1,NDP(1)+NDP(2))=YO(4)          RIN09070
DO 300 II=2,NDP(2)                    RIN09080
I=NDP(1)+NDP(2)-II+1                 RIN09090
CALL RUK(-DX(2),-DP(2),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN09100
B1(1,1,I)=YO(1)                      RIN09110
B1(2,1,I)=YO(2)                      RIN09120
B2(1,1,I)=YO(3)                      RIN09130
300 B2(2,1,I)=YO(4)                   RIN09140
CALL RUK(-DX(2)/2.,-DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN09150
CALL RUK(-DX(1)/2.,-DP(1)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN09160
B1(1,1,NDP(1))=YO(1)                 RIN09170
B1(2,1,NDP(1))=YO(2)                 RIN09180
B2(1,1,NDP(1))=YO(3)                 RIN09190
B2(2,1,NDP(1))=YO(4)                 RIN09200
DO 301 II=2,NDP(1)                   RIN09210
I=NDP(1)-II+1                        RIN09220
CALL RUK(-DX(1),-DP(1),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN09230
B1(1,1,I)=YO(1)                      RIN09240
B1(2,1,I)=YO(2)                      RIN09250
B2(1,1,I)=YO(3)                      RIN09260
301 B2(2,1,I)=YO(4)                   RIN09270
C                                       RIN09280
YO(1)=0.                              RIN09290
YO(2)=1.                              RIN09300
YO(3)=0.                              RIN09310
YO(4)=0.                              RIN09320
XNN=1.                                 RIN09330
CALL RUK(-DX(2)/2.,-DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN09340
B1(1,2,NDP(1)+NDP(2))=YO(1)          RIN09350
B1(2,2,NDP(1)+NDP(2))=YO(2)          RIN09360
B2(1,2,NDP(1)+NDP(2))=YO(3)          RIN09370
B2(2,2,NDP(1)+NDP(2))=YO(4)          RIN09380
DO 400 II=2,NDP(2)                    RIN09390
I=NDP(1)+NDP(2)-II+1                 RIN09400
CALL RUK(-DX(2),-DP(2),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN09410
B1(1,2,I)=YO(1)                      RIN09420
B1(2,2,I)=YO(2)                      RIN09430
B2(1,2,I)=YO(3)                      RIN09440
400 B2(2,2,I)=YO(4)                   RIN09450
CALL RUK(-DX(2)/2.,-DP(2)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN09460
CALL RUK(-DX(1)/2.,-DP(1)/2.,XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN09470
B1(1,2,NDP(1))=YO(1)                 RIN09480
B1(2,2,NDP(1))=YO(2)                 RIN09490
B2(1,2,NDP(1))=YO(3)                 RIN09500
B2(2,2,NDP(1))=YO(4)                 RIN09510
DO 401 II=2,NDP(1)                   RIN09520
I=NDP(1)-II+1                        RIN09530
CALL RUK(-DX(1),-DP(1),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN09540
B1(1,2,I)=YO(1)                      RIN09550
B1(2,2,I)=YO(2)                      RIN09560
B2(1,2,I)=YO(3)                      RIN09570
401 B2(2,2,I)=YO(4)                   RIN09580
C                                       RIN09590
```

```
DO 500 I=1,NDP(1)
CALL AIN(A1(1,1,I),A1I)
CALL AMU(A2(1,1,I),A1I,AT)
CALL AMU(AT,B1(1,1,I),A1(1,1,I))
DO 21 II=1,2
DO 21 JJ=1,2
21 AT(II,JJ)=B2(II,JJ,I)-A1(II,JJ,I)
CALL AIN(AT,B2(1,1,I))
C(I)=B2(1,2,I)/AL*DP(1)
D(I)=B2(2,2,I)/AL*DP(1)
500 CONTINUE
INDEX1=NDP(1)+1
INDEX2=NDP(1)+NDP(2)
DO 501 I=INDEX1,INDEX2
CALL AIN(A1(1,1,I),A1I)
CALL AMU(A2(1,1,I),A1I,AT)
CALL AMU(AT,B1(1,1,I),A1(1,1,I))
DO 22 II=1,2
DO 22 JJ=1,2
22 AT(II,JJ)=B2(II,JJ,I)-A1(II,JJ,I)
CALL AIN(AT,B2(1,1,I))
C(I)=B2(1,2,I)/AL*DP(2)
D(I)=B2(2,2,I)/AL*DP(2)
501 CONTINUE
C
RETURN
END
```

RIN09600
RIN09610
RIN09620
RIN09630
RIN09640
RIN09650
RIN09660
RIN09670
RIN09680
RIN09690
RIN09700
RIN09710
RIN09720
RIN09730
RIN09740
RIN09750
RIN09760
RIN09770
RIN09780
RIN09790
RIN09800
RIN09810
RIN09820
RIN09830
RIN09840
RIN09850
RIN09860

```

SUBROUTINE CHECK(AL,BET,EPS,EL1,NDX,NDP,IRUN,RECALC) RIN09870
C>-----RIN09880
C RIN09890
C RIN09900
C FUNCTION - CHECK INPUT VARIABLES FOR RECALCULATION RIN09910
C INFLUENCE COEFFICIENTS C,D RIN09920
C RIN09930
C RESTRICTIONS - ONLY USED FOR MULTIPLE RUNS WITH IRUN.GE.2 RIN09940
C RIN09950
C REMARKS - RIN09960
C RIN09970
C EXTERNAL REFERENCES - RIN09980
C RIN09990
C ARGUMENT DEFINITION: RIN10000
C NAME DESCRIPTION RIN10010
C AL,BET,... INPUT VARIABLES, ARE THEY CHANGED ? RIN10020
C RECALC LOGICAL VARIABLE, IF .TRUE., RECALCULATE C,D RIN10030
C RIN10040
C RIN10050
C>-----RIN10060
IMPLICIT REAL*8 (A-H,O-Z) RIN10070
COMMON/BOLD /ALOLD,BETOLD,EPSOLD,EL1OLD,NDXOLD(3),NDPOLD(3) RIN10080
DIMENSION NDX(3),NDP(3) RIN10090
LOGICAL RECALC,TEMP RIN10100
RECALC=.TRUE. RIN10110
IF(IRUN.GT.1)THEN RIN10120
  TEMP=.TRUE. RIN10130
  TEMP=TEMP.AND.(AL .EQ. ALOLD) RIN10140
  TEMP=TEMP.AND.(EPS.EQ.EPSOLD).AND.(EL1.EQ.EL1OLD) RIN10150
  DO 1 I=1,3 RIN10160
    TEMP=TEMP.AND.(NDX(I).EQ.NDXOLD(I)) RIN10170
    TEMP=TEMP.AND.(NDP(I).EQ.NDPOLD(I)) RIN10180
1  CONTINUE RIN10190
  IF(TEMP)RECALC=.FALSE. RIN10200
END IF RIN10210
ALOLD =AL RIN10220
EPSOLD=EPS RIN10230
EL1OLD=EL1 RIN10240
DO 2 I=1,3 RIN10250
  NDXOLD(I)=NDX(I) RIN10260
  NDPOLD(I)=NDP(I) RIN10270
2  CONTINUE RIN10280
RETURN RIN10290
END RIN10300

```



```

SUBROUTINE CONST2(H0,H1,PF) RIN10800
C>-----RIN10810
C RIN10820
C RIN10830
C REMARKS - SEE CONST RIN10840
C RIN10850
C RIN10860
C>-----RIN10870
IMPLICIT REAL*8 (A-H,O-Z) RIN10880
COMMON/BCD /XLOC(100),C(100),D(100),NDX(3),NDP(3),DX(3),DP(3) RIN10890
COMMON/BCOEFF/RK,RC,R1,R2,XCAV,ICAV RIN10900
COMMON/BINT /IELAS,IBACK,IPR RIN10910
R0=H1+H0 RIN10920
R1=H0 RIN10930
R2=H1-H0 RIN10940
CTEMP=(-1)**(IBACK+1) RIN10950
RK=2.*H0*H1/R0*(1.+CTEMP*H0*H1*PF) RIN10960
RC=1./R0*(1.-CTEMP*H1**2*PF) RIN10970
XCAV=0. RIN10980
IF(IBACK.EQ.1)THEN RIN10990
C IF(IPR.EQ.1)WRITE(4,*)' ' RIN11000
PMIN=100. RIN11010
DO 10 I=1,NDP(2)+NDP(3) RIN11020
C IF(IPR.EQ.1)WRITE(4,*)P(XLOC(I)) RIN11030
PMIN=DMIN1(PMIN,P(XLOC(I))) RIN11040
10 CONTINUE RIN11050
IF(PMIN.GE.0.DO)RETURN RIN11060
H11=H1-H0 RIN11070
XCAV=0.5 RIN11080
HCAV=H1+H11*(XCAV-1.) RIN11090
ICOUNT=0 RIN11100
20 ICOUNT=ICOUNT+1 RIN11110
HCVOLD=HCAV RIN11120
FHCAV=PF+1./H11*(-1./H1+HCAV/2./H1**2+1./2./HCAV) RIN11130
F1HCAV=1./H11*(1./2./H1**2-1./2./HCAV**2) RIN11140
HCAV=HCAV-FHCAV/F1HCAV RIN11150
XCAV=(HCAV-H1)/H11+1. RIN11160
C IF(IPR.EQ.1)WRITE(4,*)XCAV RIN11170
IF(ICOUNT.LT.10.AND.DABS(HCAV-HCVOLD).GT.1.D-7)GOTO 20 RIN11180
RK=HCAV RIN11190
RC=1./2./HCAV RIN11200
END IF RIN11210
RETURN RIN11220
END RIN11230

```

```
      SUBROUTINE DFN1(X,Y,D)
C>-----RIN11240
C      RIN11250
C      RIN11260
C      RIN11270
C      FUNCTION          - DERIVATIVE ROUTINE USED BY RUK
C                        (NO HYDRODYNAMICS)
C                        RIN11280
C                        RIN11290
C                        RIN11300
C      RESTRICTIONS      -
C                        RIN11310
C                        RIN11320
C      REMARKS           - NOTE VARIABLES PASSED IN COMMON
C                        RIN11330
C                        RIN11340
C      EXTERNAL REFERENCES -
C                        RIN11350
C                        RIN11360
C      ARGUMENT DEFINITION:
C      NAME              DESCRIPTION
C      X                  VALUE OF X FOR DERIVATIVES
C                        RIN11370
C                        RIN11380
C      Y                  VALUES OF INDEPENDENT VARIABLES
C                        RIN11390
C                        RIN11400
C      D                  VALUES OF DERIVATIVES
C                        RIN11410
C                        RIN11420
C                        RIN11430
C>-----RIN11440
      IMPLICIT REAL*8 (A-H,O-Z)
      DIMENSION Y(4),D(4)
      COMMON/BPAR /AL,BET,PO,EPS,PF,S,U,DT
      D(1)=Y(2)
      D(2)=Y(3)
      D(3)=Y(4)
      D(4)=-Y(1)/AL
      RETURN
      END
      RIN11450
      RIN11460
      RIN11470
      RIN11480
      RIN11490
      RIN11500
      RIN11510
      RIN11520
      RIN11530
```

```

SUBROUTINE DFN2(X,Y,D)
C>-----RIN11540
C RIN11550
C RIN11560
C RIN11570
C FUNCTION - DERIVATIVE ROUTINE USED BY RUK RIN11580
C (HYDRODYNAMICS INCLUDED) RIN11590
C RIN11600
C RESTRICTIONS - RIN11610
C RIN11620
C REMARKS - NOTE VARIABLES PASSED IN COMMON RIN11630
C RIN11640
C EXTERNAL REFERENCES - P RIN11650
C RIN11660
C ARGUMENT DEFINITION: RIN11670
C NAME DESCRIPTION RIN11680
C X VALUE OF X FOR DERIVATIVES RIN11690
C Y VALUES OF INDEPENDENT VARIABLES RIN11700
C D VALUES OF DERIVATIVES RIN11710
C RIN11720
C RIN11730
C>-----RIN11740
IMPLICIT REAL*8 (A-H,O-Z) RIN11750
DIMENSION Y(4),D(4) RIN11760
COMMON/BPAR /AL,BET,PO,EPS,PF,S,U,DT RIN11770
COMMON/BFLAG /IFLAG,IPLOAD RIN11780
D(1)=Y(2) RIN11790
D(2)=Y(3) RIN11800
D(3)=Y(4) RIN11810
H=0 RIN11820
IF(IPLOAD.EQ.1)H=PO RIN11830
IF(X.GT.0.)H=H-P(X) RIN11840
H=H*BET RIN11850
D(4)=(H-Y(1))/AL RIN11860
RETURN RIN11870
END RIN11880

```

```

SUBROUTINE DFN3(X,Y,D)
C>-----RIN11890
C RIN11900
C RIN11910
C RIN11920
C FUNCTION - DERIVATIVE ROUTINE USED BY RUK RIN11930
C (HYDRODYNAMICS INCLUDED, NO PRELOAD P0) RIN11940
C RIN11950
C RESTRICTIONS - RIN11960
C RIN11970
C REMARKS - NOTE VARIABLES PASSED IN COMMON RIN11980
C RIN11990
C EXTERNAL REFERENCES - P RIN12000
C RIN12010
C ARGUMENT DEFINITION: RIN12020
C NAME DESCRIPTION RIN12030
C X VALUE OF X FOR DERIVATIVES RIN12040
C Y VALUES OF INDEPENDENT VARIABLES RIN12050
C D VALUES OF DERIVATIVES RIN12060
C RIN12070
C RIN12080
C>-----RIN12090
IMPLICIT REAL*8 (A-H,O-Z) RIN12100
DIMENSION Y(4),D(4) RIN12110
COMMON/BPAR /AL,BET,P0,EPS,PF,S,U,DT RIN12120
COMMON/BINT /IELAS,IBACK,IPR RIN12130
D(1)=Y(2) RIN12140
D(2)=Y(3) RIN12150
D(3)=Y(4) RIN12160
H=BET*P0 RIN12170
D(4)=(H-Y(1))/AL RIN12180
RETURN RIN12190
END RIN12200
```



```

CHI=YO(1)-1.0
YO(1)=1.0
B=CHI/(W2(3)*W1(1)/W1(3)-W2(1))
A=-B*W2(3)/W1(3)
YO(2)=A*W1(2)+B*W2(2)+YO(2)
YO(3)=0.
YO(4)=A*W1(4)+B*W2(4)
FORCE=YO(4)*AL/BET
HELAS=1.0-YO(1)
H1ELAS=-YO(2)
H2ELAS=-YO(3)
H3ELAS=-YO(4)
WRITE(6,'(F8.4,5F10.5,F13.5)')1.0,HELAS,H1ELAS,H2ELAS
$,H3ELAS,0.,FORCE
END IF
RETURN
END

```

SUBROUTINE ERRMSG

```

C-----
C
C
C   FUNCTION          - PRINT ERROR TERMS UPON EXIT FROM *ZSCNT*
C
C   RESTRICTIONS     - CALLED ONLY IF IER.NE.0 FROM *ZSCNT*
C
C   REMARKS          - FINAL VALUES OF X AND F FROM EVAL
C                   ARE PRINTED
C
C   INPUT/OUTPUT:
C   UNIT      DESCRIPTION
C     4        TERMINAL OUTPUT
C     6        OUTPUT FILE
C-----
C>
IMPLICIT REAL*8 (A-H,O-Z)
COMMON/BLAST /X1,X2,F1,F2
WRITE(4,'(8X,'W(1)',10X,'W(2)',10X,'F(1)',10X,'F(2)''
+ ',/)'')
WRITE(4,'(1X,4(2X,E12.5))')X1,X2,F1,F2
WRITE(6,'(8X,'W(1)',10X,'W(2)',10X,'F(1)',10X,'F(2)''
+ ')')
WRITE(6,'(1X,4(2X,E12.5))')X1,X2,F1,F2
RETURN
END

```

```

SUBROUTINE EVAL(X,F,N,PAR)
C>-----RIN13210
C RIN13220
C RIN13230
C RIN13240
C FUNCTION - DEFINES EQUATIONS FOR H AND H' WHICH ARE RIN13250
C SOLVED BY SECANT METHOD *ZSCNT* RIN13260
C RIN13270
C RESTRICTIONS - RIN13280
C RIN13290
C REMARKS - NOTE VARIABLE PASSED IN COMMON RIN13300
C RIN13310
C EXTERNAL REFERENCES - CONST RIN13320
C RIN13330
C INPUT/OUTPUT: RIN13340
C UNIT DESCRIPTION RIN13350
C RIN13360
C ARGUMENT DEFINITION: RIN13370
C NAME DESCRIPTION RIN13380
C X X(1) IS DISPLACEMENT AT 1 RIN13390
C X(2) IS X' AT 1 RIN13400
C F EQUATIONS (2) FOR H AND H' RIN13410
C N NO. OF EQUATIONS (N=2) RIN13420
C PAR NOT USED RIN13430
C RIN13440
C RIN13450
C>-----RIN13460
IMPLICIT REAL*8 (A-H,O-Z) RIN13470
DIMENSION X(N),F(N),PAR(1) RIN13480
COMMON/BPAR /AL,BET,PO,EPS,PF RIN13490
COMMON/BCD /XLOC(100),C(100),D(100),NDX(3),NDP(3),DX(3),DP(3) RIN13500
COMMON/BCOEFF/RK,RC,R1,R2,XCAV,ICAV RIN13510
COMMON/BINT /IELAS,IBACK,IPR RIN13520
COMMON/BLAST /X1LAST,X2LAST,F1LAST,F2LAST RIN13530
COMMON/BPROF /C1 RIN13540
COMMON/BFLAG /IFLAG,IPLOAD RIN13550
H1=1.0-X(1) RIN13560
H11=C1-X(2) RIN13570
H0=H1-H11 RIN13580
CALL CONST(H0,H1,PF) RIN13590
SUM1=X(1) RIN13600
SUM2=X(2) RIN13610
INDEX=0 RIN13620
5 CONTINUE RIN13630
INDEX=INDEX+1 RIN13640
IF(IFLAG.EQ.1.AND.INDEX.GT.NDP(2)+NDP(3))GOTO 6 RIN13650
IF(IFLAG.EQ.2.AND.INDEX.GT.NDP(2))GOTO 6 RIN13660
IF(XLOC(INDEX).LT.DMIN1(1.-EPS,XCAV))GOTO 5 RIN13670
IF(XLOC(INDEX).LT.XCAV)THEN RIN13680
PRES=0 RIN13690
ELSE RIN13700
PRES=P(XLOC(INDEX)) RIN13710
END IF RIN13720
IF(XLOC(INDEX).GT.1.-EPS) PRES=PRES-PO RIN13730
7 CONTINUE RIN13740
SUM1=SUM1+C(INDEX)*PRES*BET RIN13750

```

	SUM2=SUM2+D(INDEX)*PRES*BET	RIN13760
	GOTO 5	RIN13770
6	CONTINUE	RIN13780
	F(1)=SUM1	RIN13790
	F(2)=SUM2	RIN13800
	X1LAST=X(1)	RIN13810
	X2LAST=X(2)	RIN13820
	F1LAST=F(1)	RIN13830
	F2LAST=F(2)	RIN13840
	RETURN	RIN13850
	END	RIN13860

```

SUBROUTINE EVAL3(X,F,N,PAR)                                RIN13870
C>-----RIN13880
C                                                    RIN13890
C                                                    RIN13900
C    FUNCTION          - DEFINES EQUATIONS FOR H AND H' WHICH ARE RIN13910
C                      SOLVED BY SECANT METHOD *ZSCNT*          RIN13920
C                                                    RIN13930
C    RESTRICTIONS      -                                       RIN13940
C                                                    RIN13950
C    REMARKS           - NOTE VARIABLE PASSED IN COMMON        RIN13960
C                                                    RIN13970
C    EXTERNAL REFERENCES - CONST                                RIN13980
C                                                    RIN13990
C    INPUT/OUTPUT:                                           RIN14000
C    UNIT              DESCRIPTION                             RIN14010
C                                                    RIN14020
C    ARGUMENT DEFINITION:                                     RIN14030
C    NAME              DESCRIPTION                             RIN14040
C    X                  X(1) IS DISPLACEMENT AT 1             RIN14050
C                      X(2) IS X' AT 1                        RIN14060
C    F                  EQUATIONS (2) FOR H AND H'            RIN14070
C    N                  NO. OF EQUATIONS (N=2)                 RIN14080
C    PAR                NOT USED                                RIN14090
C                                                    RIN14100
C>-----RIN14110
IMPLICIT REAL*8 (A-H,O-Z)                                RIN14130
DIMENSION X(N),F(N),PAR(1)                                RIN14140
COMMON/BPAR /AL,BET,PO,EPS,PF                               RIN14150
COMMON/BCD /XLOC(100),C(100),D(100),NDX(3),NDP(3),DX(3),DP(3) RIN14160
COMMON/BCOEFF/RK,RC,R1,R2,XCAV,ICAV                         RIN14170
COMMON/BINT /IELAS,IBACK,IPR                                RIN14180
COMMON/BLAST /X1LAST,X2LAST,F1LAST,F2LAST                   RIN14190
COMMON/BPROF /C1                                             RIN14200
COMMON/BFLAG /IFLAG,IPLOAD                                  RIN14210
H1=1.0-X(1)                                                 RIN14220
H11=C1-X(2)                                                 RIN14230
H0=H1-H11                                                   RIN14240
CALL CONST(H0,H1,PF)                                        RIN14250
SUM1=X(1)                                                   RIN14260
SUM2=X(2)                                                   RIN14270
INDEX=0                                                     RIN14280
5 CONTINUE                                                  RIN14290
INDEX=INDEX+1                                              RIN14300
IF(INDEX.GT.NDP(2)+NDP(3))GOTO 6                            RIN14310
IF(XLOC(INDEX).LT.DMIN1(1.-EPS,XCAV))GOTO 5                 RIN14320
IF(XLOC(INDEX).LT.XCAV)THEN                                  RIN14330
  PRES=0                                                    RIN14340
ELSE                                                         RIN14350
  PRES=P(XLOC(INDEX))                                       RIN14360
END IF                                                       RIN14370
PRES=PRES-PO                                               RIN14380
7 CONTINUE                                                  RIN14390
SUM1=SUM1+C(INDEX)*PRES*BET                                 RIN14400
SUM2=SUM2+D(INDEX)*PRES*BET                                 RIN14410

```

	GOTO 5	RIN14420
6	CONTINUE	RIN14430
	F(1)=SUM1	RIN14440
	F(2)=SUM2	RIN14450
	X1LAST=X(1)	RIN14460
	X2LAST=X(2)	RIN14470
	F1LAST=F(1)	RIN14480
	F2LAST=F(2)	RIN14490
	RETURN	RIN14500
	END	RIN14510

```

SUBROUTINE EVAL4(X,F,N,PAR)
C-----
C
C
C   FUNCTION          -  DEFINES EQUATIONS FOR H AND H' WHICH ARE
C                       SOLVED BY SECANT METHOD *ZSCNT*
C
C   RESTRICTIONS     -
C
C   REMARKS          -  NOTE VARIABLE PASSED IN COMMON
C
C   EXTERNAL REFERENCES -  CONST
C
C   INPUT/OUTPUT:
C   UNIT            DESCRIPTION
C
C   ARGUMENT DEFINITION:
C   NAME            DESCRIPTION
C   X                X(1) IS DISPLACEMENT AT 1
C                   X(2) IS X' AT 1
C   F                EQUATIONS (2) FOR H AND H'
C   N                NO. OF EQUATIONS (N=2)
C   PAR             NOT USED
C-----
C>
IMPLICIT REAL*8 (A-H,O-Z)
DIMENSION X(N),F(N),PAR(1)
COMMON/BPAR /AL,BET,P0,EPS,PF
COMMON/BCD /XLOC(100),C(100),D(100),NDX(3),NDP(3),DX(3),DP(3)
COMMON/BCOEFF/RK,RC,R1,R2,XCAV,ICAV
COMMON/BINT /IELAS,IBACK,IPR
COMMON/BLAST /X1LAST,X2LAST,F1LAST,F2LAST
COMMON/BPROF /C1
COMMON/BFLAG /IFLAG,IPLOAD
H1=1.0-X(1)
H11=C1-X(2)
H0=H1-H11
CALL CONST(H0,H1,PF)
SUM1=X(1)
SUM2=X(2)
INDEX=0
5 CONTINUE
INDEX=INDEX+1
IF(INDEX.GT.NDP(1)+NDP(2))GOTO 6
IF(XLOC(INDEX).LT.DMIN1(1.-EPS,XCAV))GOTO 5
IF(XLOC(INDEX).LT.XCAV)THEN
PRES=0
ELSE
PRES=P(XLOC(INDEX))
END IF
PRES=PRES-P0
7 CONTINUE
SUM1=SUM1+C(INDEX)*PRES*BET
SUM2=SUM2+D(INDEX)*PRES*BET

```

	GOTO 5	RIN15070
6	CONTINUE	RIN15080
	F(1)=SUM1	RIN15090
	F(2)=SUM2	RIN15100
	X1LAST=X(1)	RIN15110
	X2LAST=X(2)	RIN15120
	F1LAST=F(1)	RIN15130
	F2LAST=F(2)	RIN15140
	RETURN	RIN15150
	END	RIN15160
	FUNCTION P(PSI)	RIN15170
C>	-----	RIN15180
C		RIN15190
C		RIN15200
C	FUNCTION - RETURN HYDRODYNAMIC PRESSURE AT PSI	RIN15210
C		RIN15220
C	RESTRICTIONS -	RIN15230
C		RIN15240
C	REMARKS - NOTE VARIABLES PASSED IN COMMON	RIN15250
C		RIN15260
C	EXTERNAL REFERENCES - DABS	RIN15270
C		RIN15280
C	ARGUMENT DEFINITION:	RIN15290
C	NAME DESCRIPTION	RIN15300
C	PSI VALUE OF X WHERE PRESSURE IS DESIRED	RIN15310
C		RIN15320
C		RIN15330
C>	-----	RIN15340
	IMPLICIT REAL*8 (A-H,O-Z)	RIN15350
	COMMON/BCOEFF/RK,RC,R1,R2,XCAV,ICAV	RIN15360
	COMMON/BINT /IELAS,IBACK,IPR	RIN15370
	H=R1+R2*PSI	RIN15380
	HINV=1./H	RIN15390
	P=0.	RIN15400
	IF(PSI.GE.XCAV.AND.DABS(R2).GT.1.D-10)THEN	RIN15410
	P=1./R2*(HINV*(-1.+RK/2.*HINV)+RC)	RIN15420
	END IF	RIN15430
	IF(IBACK.EQ.1)THEN	RIN15440
	P=-P	RIN15450
	END IF	RIN15460
	RETURN	RIN15470
	END	RIN15480

```

SUBROUTINE PRT(X,Y,ITERM)                                RIN15490
C>-----RIN15500
C                                                    RIN15510
C                                                    RIN15520
C    FUNCTION          - USING DISPLACEMENTS, CONVERT TO FILM  RIN15530
C                      THICKNESSES AND PRINT OUT              RIN15540
C                                                    RIN15550
C    RESTRICTIONS      -                                       RIN15560
C                                                    RIN15570
C    REMARKS           -                                       RIN15580
C                                                    RIN15590
C    EXTERNAL REFERENCES - P                                     RIN15600
C                                                    RIN15610
C    INPUT/OUTPUT:                                           RIN15620
C    UNIT              DESCRIPTION                             RIN15630
C    ITERM             OUTPUT UNIT                             RIN15640
C                                                    RIN15650
C    ARGUMENT DEFINITION:                                     RIN15660
C    NAME              DESCRIPTION                             RIN15670
C    X                  DIMENSIONLESS LENGTH VARIABLE         RIN15680
C    Y                  DISPLACEMENT AND ITS DERIVATIVES      RIN15690
C    ITERM              OUTPUT UNIT                             RIN15700
C                                                    RIN15710
C>-----RIN15720
IMPLICIT REAL*8 (A-H,O-Z)                                RIN15730
COMMON/BINT /IELAS,IBACK,IPR                               RIN15740
COMMON/BPROF /C1                                           RIN15750
DIMENSION Y(4)                                             RIN15770
H=1.-Y(1)+C1*(X-1.)                                       RIN15780
H1=C1-Y(2)                                                 RIN15790
H2=-Y(3)                                                   RIN15800
H3=-Y(4)                                                   RIN15810
PRES=0.                                                    RIN15820
IF(IELAS.EQ.0.AND.X.GE.0.)PRES=P(X)                       RIN15830
WRITE(ITERM,5)X,H,H1,H2,H3,PRES                           RIN15840
5  FORMAT(F10.4,5F12.5)                                     RIN15850
RETURN                                                    RIN15860
END                                                        RIN15870

```

```

SUBROUTINE PRTOU(XX)
C-----
C
C
C      FUNCTION          - PRINT FILM THICK., PRES., ETC.
C
C      RESTRICTIONS      -
C
C      REMARKS           - NOTE VARIABLES PASSED IN COMMON
C
C      EXTERNAL REFERENCES - DFN1,DFN2,DFN3,PRT,RUK
C
C      INPUT/OUTPUT:
C      UNIT      DESCRIPTION
C      6          OUTPUT FILE
C
C      ARGUMENT DEFINITION:
C      NAME      DESCRIPTION
C      XX(2)     DISPLACEMENT, SLOPE OF DISPLACEMENT
C-----
C
C      IMPLICIT REAL*8 (A-H,O-Z)
COMMON/BINT /IELAS,IBACK,IPR
COMMON/BELAS /FORCE,HELAS,H1ELAS,H2ELAS,H3ELAS,W1(4),W2(4)
COMMON/BCD   /XLOC(100),C(100),D(100),NDX(3),NDP(3),DX(3),DP(3)
COMMON/BFLAG /IFLAG,IPLOAD
DIMENSION   XX(2),YO(4),YT(4),DJ(4),CKJ(4,4)
EXTERNAL    DFN1,DFN2,DFN3
C
C      XNN=1.0
YO(1)=XX(1)
YO(2)=XX(2)
YO(3)=0.
YO(4)=0.
IF(IBACK.EQ.1.AND.FORCE.GT.1.D-8)THEN
  WRITE(6,1009)
  WRITE(6,'(F8.4,5F10.5,F13.5)')1.0,HELAS,H1ELAS,H2ELAS
$      ,H3ELAS,0.,FORCE
RETURN
ELSE
  WRITE(6,1007)
  IF(IPR.EQ.1)THEN
    WRITE(6,'(''@'')')
  ELSE
    WRITE(6,'('' ''')')
  END IF
  CALL PRT(XNN,YO,6)
END IF
IF(IPR.EQ.1)THEN
  IF(IFLAG.EQ.2.OR.IFLAG.EQ.4)GOTO 111
  IPLOAD=1
  CALL RUK(-DX(3)/2.,-DP(3)/2.,XNN,XNN,YO,YO,4,DFN2,YT,DJ,CKJ,4)
  CALL PRT(XNN,YO,6)
  DO 9 JJ=2,NDP(3)

```

```
J=NDP(3)+1-JJ                                RIN16430
CALL RUK(-DX(3),-DP(3),XNN,XNN,YO,YO,4,DFN2,YT,DJ,CKJ,4) RIN16440
CALL PRT(XNN,YO,6)                             RIN16450
9 CONTINUE                                     RIN16460
CALL RUK(-DX(3)/2.,-DP(3)/2.,XNN,XNN,YO,YO,4,DFN2,YT,DJ,CKJ,4) RIN16470
CALL PRT(XNN,YO,6)                             RIN16480
111 CONTINUE                                  RIN16490
IF(IFLAG.EQ.1)THEN                             RIN16500
  IPLOAD=0                                       RIN16510
ELSE IF(IFLAG.EQ.2)THEN                         RIN16520
  IPLOAD=1                                       RIN16530
ELSE IF(IFLAG.EQ.3)THEN                         RIN16540
  IPLOAD=1                                       RIN16550
ELSE IF(IFLAG.EQ.4)THEN                         RIN16560
  IPLOAD=1                                       RIN16570
END IF                                           RIN16580
CALL RUK(-DX(2)/2.,-DP(2)/2.,XNN,XNN,YO,YO,4,DFN2,YT,DJ,CKJ,4) RIN16590
CALL PRT(XNN,YO,6)                             RIN16600
DO 8 JJ=2,NDP(2)                               RIN16610
J=NDP(2)+1-JJ                                  RIN16620
CALL RUK(-DX(2),-DP(2),XNN,XNN,YO,YO,4,DFN2,YT,DJ,CKJ,4) RIN16630
CALL PRT(XNN,YO,6)                             RIN16640
8 CONTINUE                                     RIN16650
CALL RUK(-DX(2)/2.,-DP(2)/2.,XNN,XNN,YO,YO,4,DFN2,YT,DJ,CKJ,4) RIN16660
CALL PRT(XNN,YO,6)                             RIN16670
IF(IFLAG.EQ.1.OR.IFLAG.EQ.2)THEN              RIN16680
  WRITE(6,('( '@'))' )                          RIN16690
  RETURN                                         RIN16700
END IF                                           RIN16710
DO 7 J=1,NDP(1)                                RIN16720
IF(IFLAG.EQ.4)THEN                             RIN16730
  CALL RUK(-DX(1),-DP(1),XNN,XNN,YO,YO,4,DFN3,YT,DJ,CKJ,4) RIN16740
ELSE                                             RIN16750
  CALL RUK(-DX(1),-DP(1),XNN,XNN,YO,YO,4,DFN1,YT,DJ,CKJ,4) RIN16760
END IF                                           RIN16770
CALL PRT(XNN,YO,6)                             RIN16780
7 CONTINUE                                     RIN16790
END IF                                           RIN16800
WRITE(6,('( '@'))' )                          RIN16810
RETURN                                         RIN16820
C                                               RIN16830
1007 FORMAT(7X,1HX,10X,1HH,11X,2HH',9X,3HH'',9X,4HH''',8X,4HPRES) RIN16840
1009 FORMAT(5X,1HX,8X,1HH,9X,2HH',7X,3HH'',7X,4HH''',6X,4HPRES RIN16850
$,8X,5HFORCE)                                RIN16860
C                                               RIN16870
END                                           RIN16880
```

```

SUBROUTINE RUK(DX,DP,XO,XN,YO,YN,NO,DFN,YT,D,CK,ID)
C-----RIN16890
C-----RIN16900
C-----RIN16910
C-----RIN16920
C  FUNCTION          - RUNGE-KUTTA INTEGRATION      RIN16930
C-----RIN16940
C  RESTRICTIONS      -                               RIN16950
C-----RIN16960
C  REMARKS           -                               RIN16970
C-----RIN16980
C  EXTERNAL REFERENCES - DFN                        RIN16990
C-----RIN17000
C  ARGUMENT DEFINITION:                            RIN17010
C  NAME              DESCRIPTION                    RIN17020
C-----RIN17030
C-----RIN17040
C-----RIN17050
  IMPLICIT REAL*8 (A-H,O-Z)                        RIN17060
  DIMENSION YO(ID),YN(ID),D(ID),YT(ID),CK(4,ID)   RIN17070
  N=DP/DX+.001                                     RIN17080
  XN=XO                                             RIN17090
  DO 2 J=1,NO                                       RIN17100
2  YN(J)=YO(J)                                       RIN17110
  DO 5 I=1,N                                         RIN17120
  CALL DFN(XN,YN,D)                                  RIN17130
  DO 6 J=1,NO                                       RIN17140
  CK(1,J)=D(J)*DX                                   RIN17150
6  YT(J)=YN(J)+CK(1,J)/2.                            RIN17160
  CALL DFN(XN+DX/2.,YT,D)                            RIN17170
  DO 7 J=1,NO                                       RIN17180
  CK(2,J)=D(J)*DX                                   RIN17190
7  YT(J)=YN(J)+CK(2,J)/2.                            RIN17200
  CALL DFN(XN+DX/2.,YT,D)                            RIN17210
  DO 8 J=1,NO                                       RIN17220
  CK(3,J)=D(J)*DX                                   RIN17230
8  YT(J)=YN(J)+CK(3,J)                              RIN17240
  CALL DFN(XN+DX,YT,D)                                RIN17250
  DO 9 J=1,NO                                       RIN17260
  CK(4,J)=D(J)*DX                                   RIN17270
  YN(J)=YN(J)+(CK(1,J)+2.*(CK(2,J)+CK(3,J))+CK(4,J))/6. RIN17280
9  CONTINUE                                          RIN17290
5  XN=XN+DX                                         RIN17300
  RETURN                                             RIN17310
  END.                                              RIN17320

```

```

SUBROUTINE ZSCNT (FCN,NSIG,N,ITMAX,PAR,X,FNORM,WK,IER)          ZSC00010
C  IMSL ROUTINE NAME - ZSCNT                                   ZSC00020
C  THIS PROGRAM SHOULD BE COMPILED IN FORTRAN 4 (FORTHX)      ZSC00030
C  -----ZSC00040
C  ZSC00050
C  COMPUTER - IBM/DOUBLE                                       ZSC00060
C  ZSC00070
C  LATEST REVISION - JUNE 1, 1980                             ZSC00080
C  ZSC00090
C  PURPOSE - SOLVE A SYSTEM OF NONLINEAR EQUATIONS           ZSC00100
C  ZSC00110
C  USAGE - CALL ZSCNT (FCN,NSIG,N,ITMAX,PAR,X,FNORM,         ZSC00120
C  WK,IER)                                                    ZSC00130
C  ZSC00140
C  ARGUMENTS FCN - THE NAME OF A USER-SUPPLIED SUBROUTINE WHICH ZSC00150
C  EVALUATES THE SYSTEM OF EQUATIONS TO BE                   ZSC00160
C  SOLVED. FCN MUST BE DECLARED EXTERNAL IN                 ZSC00170
C  THE CALLING PROGRAM AND MUST HAVE THE                   ZSC00180
C  FOLLOWING FORM,                                          ZSC00190
C  SUBROUTINE FCN(X,F,N,PAR)                                ZSC00200
C  DIMENSION X(N),F(N),PAR(1)                              ZSC00210
C  F(1)=                                                    ZSC00220
C  .                                                         ZSC00230
C  F(N)=                                                    ZSC00240
C  RETURN                                                  ZSC00250
C  END                                                    ZSC00260
C  GIVEN X(1)...X(N), FCN MUST EVALUATE THE                ZSC00270
C  FUNCTIONS F(1)...F(N) WHICH ARE TO BE MADE              ZSC00280
C  ZERO. X SHOULD NOT BE ALTERED BY FCN. THE              ZSC00290
C  PARAMETERS IN VECTOR PAR (SEE ARGUMENT                 ZSC00300
C  PAR BELOW) MAY ALSO BE USED IN THE                    ZSC00310
C  CALCULATION OF F(1)...F(N).                            ZSC00320
C  NSIG - THE NUMBER OF DIGITS OF ACCURACY DESIRED        ZSC00330
C  IN THE COMPUTED ROOT (INPUT).                          ZSC00340
C  N - THE NUMBER OF EQUATIONS TO BE SOLVED AND          ZSC00350
C  THE NUMBER OF UNKNOWNNS (INPUT).                      ZSC00360
C  ITMAX - THE MAXIMUM ALLOWABLE NUMBER OF ITERATIONS    ZSC00370
C  (INPUT).                                               ZSC00380
C  PAR - PAR CONTAINS A PARAMETER SET WHICH IS          ZSC00390
C  PASSED TO THE USER SUPPLIED FUNCTION FCN.           ZSC00400
C  PAR MAY BE USED TO PASS ANY AUXILIARY                ZSC00410
C  PARAMETERS NECESSARY FOR COMPUTATION OF              ZSC00420
C  THE FUNCTION FCN. (INPUT)                             ZSC00430
C  X - A VECTOR OF LENGTH N. (INPUT/OUTPUT) ON INPUT,    ZSC00440
C  X IS THE INITIAL APPROXIMATION TO THE ROOT.          ZSC00450
C  ON OUTPUT, X IS THE BEST APPROXIMATION TO            ZSC00460
C  THE ROOT FOUND BY ZSCNT.                              ZSC00470
C  FNORM - ON OUTPUT, FNORM IS EQUAL TO                 ZSC00480
C  F(1)**2+...F(N)**2 AT THE POINT X.                   ZSC00490
C  WK - WORK VECTOR OF LENGTH (N+1)*(3*N+8)             ZSC00500
C  IER - ERROR PARAMETER. (OUTPUT)                      ZSC00510
C  TERMINAL ERROR                                        ZSC00520
C  IER = 129 INDICATES THAT ZSCNT FAILED TO             ZSC00530
C  CONVERGE WITHIN ITMAX ITERATIONS. THE                ZSC00540
C  USER MAY INCREASE ITMAX OR TRY A NEW                ZSC00550

```

```

C          INITIAL GUESS.                                ZSC00560
C          IER = 130 INDICATES THE ALGORITHM WAS        ZSC00570
C          UNABLE TO IMPROVE ON THE RETURNED VALUE     ZSC00580
C          OF X. THIS SITUATION ARISES WHEN THE        ZSC00590
C          SOLUTION CANNOT BE DETERMINED TO NSIG      ZSC00600
C          DIGITS DUE TO ERRORS IN THE FUNCTION       ZSC00610
C          VALUES. IT MAY ALSO INDICATE THAT THE     ZSC00620
C          ROUTINE IS TRAPPED IN THE AREA OF A        ZSC00630
C          LOCAL MINIMUM. THE USER MAY TRY A NEW     ZSC00640
C          INITIAL GUESS.                              ZSC00650
C                                                    ZSC00660
C PRECISION/HARDWARE - SINGLE AND DOUBLE/H32        ZSC00670
C                                                    ZSC00680
C          - SINGLE/H36,H48,H60                      ZSC00690
C                                                    ZSC00700
C REQD. IMSL ROUTINES - SINGLE/GGUBFS,LEQT2F,LUDATF,LUELMF,LUREFF, ZSC00700
C          UERSET,UERTST,UGETIO,ZSCNU                ZSC00710
C                                                    ZSC00720
C          - DOUBLE/GGUBFS,LEQT2F,LUDATF,LUELMF,LUREFF, ZSC00720
C          UERSET,UERTST,UGETIO,VXADD,VXMUL,VXSTO,   ZSC00730
C          ZSCNU                                      ZSC00740
C                                                    ZSC00750
C NOTATION - INFORMATION ON SPECIAL NOTATION AND     ZSC00760
C          CONVENTIONS IS AVAILABLE IN THE MANUAL    ZSC00770
C          INTRODUCTION OR THROUGH IMSL ROUTINE UHELP ZSC00780
C                                                    ZSC00790
C COPYRIGHT - 1980 BY IMSL, INC. ALL RIGHTS RESERVED. ZSC00800
C                                                    ZSC00810
C WARRANTY - IMSL WARRANTS ONLY THAT IMSL TESTING HAS BEEN ZSC00820
C          APPLIED TO THIS CODE. NO OTHER WARRANTY,  ZSC00830
C          EXPRESSED OR IMPLIED, IS APPLICABLE.     ZSC00840
C                                                    ZSC00850
C-----ZSC00860
C                                                    ZSC00870
C SUBROUTINE ZSCNT (FCN,NSIG,N,ITMAX,PAR,X,FNORM,WK,IER) ZSC00880
C          SPECIFICATIONS FOR ARGUMENTS              ZSC00890
C          INTEGER IER,ITMAX,N,NSIG                  ZSC00900
C          DOUBLE PRECISION FNORM,PAR(1),WK(1),X(N)  ZSC00910
C          SPECIFICATIONS FOR LOCAL VARIABLES        ZSC00920
C          INTEGER I1,I2,I3,I4,I5,LNEW,LOLD,N1      ZSC00930
C          EXTERNAL FCN                              ZSC00940
C          FIRST EXECUTABLE STATEMENT                ZSC00950
C          N1=N+1                                     ZSC00960
C          I1 = N1*N1+1                               ZSC00970
C          I2 = I1+N*N1                              ZSC00980
C          I3 = I2+N1                                 ZSC00990
C          I4 = I3+N1                                 ZSC01000
C          I5 = I4+N1                                 ZSC01010
C          CALL UERSET(0,LOLD)                        ZSC01020
C          CALL ZSCNU(X,N,FCN,NSIG,N1,WK(1),WK(I1),WK(I2),WK(I3),WK(I4) ZSC01030
C          * ,WK(I5),ITMAX,PAR,IER)                  ZSC01040
C          CALL FCN(X,WK,N,PAR)                       ZSC01050
C          FNORM = 0.0D0                              ZSC01060
C          DO 5 I = 1,N                               ZSC01070
C             FNORM = FNORM + WK(I)*WK(I)            ZSC01080
5 CONTINUE                                          ZSC01090
C          CALL UERSET(LOLD,LNEW)                    ZSC01100

```

IF (IER.EQ.0) GO TO 9005	ZSC01110
9000 CONTINUE	ZSC01120
CALL UERTST(IER,6HZSCNT)	ZSC01130
9005 RETURN	ZSC01140
END	ZSC01150
FUNCTION GGUBFS (DSEED)	ZSC01160
DOUBLE PRECISION DSEED	ZSC01170
DOUBLE PRECISION D2P31M,D2P31	ZSC01180
DATA D2P31M/2147483647.DO/	ZSC01190
DATA D2P31 /2147483648.DO/	ZSC01200
DSEED = DMOD(16807.DO*DSEED,D2P31M)	ZSC01210
GGUBFS = DSEED / D2P31	ZSC01220
RETURN	ZSC01230
END	ZSC01240

```
SUBROUTINE LEQT2F (A,M,N,IA,B, IDGT,WKAREA, IER)          ZSC01250
DIMENSION          A(IA,1),B(IA,1),WKAREA(1)            ZSC01260
DOUBLE PRECISION  A,B,WKAREA,D1,D2,WA                  ZSC01270
IER=0                                                     ZSC01280
JER=0                                                     ZSC01290
J = N*N+1                                                ZSC01300
K = J+N                                                  ZSC01310
MM = K+N                                                 ZSC01320
KK = 0                                                   ZSC01330
MM1 = MM-1                                              ZSC01340
JJ=1                                                     ZSC01350
DO 5 L=1,N                                              ZSC01360
  DO 5 I=1,N                                            ZSC01370
    WKAREA(JJ)=A(I,L)                                  ZSC01380
    JJ=JJ+1                                             ZSC01390
5 CONTINUE                                              ZSC01400
CALL LUDATF (WKAREA(1),A,N,N, IDGT,D1,D2,WKAREA(J),WKAREA(K),
1  WA, IER)                                             ZSC01420
IF (IER.GT.128) GO TO 25                                ZSC01430
IF (IDGT .EQ. 0 .OR. IER .NE. 0) KK = 1                ZSC01440
DO 15 I = 1,M                                          ZSC01450
  CALL LUELMF (A,B(1,I),WKAREA(J),N,N,WKAREA(MM))     ZSC01460
  IF (KK .NE. 0)                                       ZSC01470
* CALL LUREFF (WKAREA(1),B(1,I),A,WKAREA(J),N,N,WKAREA(MM), IDGT,
* WKAREA(K),WKAREA(K),JER)                            ZSC01480
  DO 10 II=1,N                                         ZSC01500
    B(II,I) = WKAREA(MM1+II)                          ZSC01510
10 CONTINUE                                             ZSC01520
IF (JER.NE.0) GO TO 20                                 ZSC01530
15 CONTINUE                                             ZSC01540
GO TO 25                                               ZSC01550
20 IER = 131                                           ZSC01560
25 JJ=1                                                ZSC01570
DO 30 J = 1,N                                          ZSC01580
  DO 30 I = 1,N                                        ZSC01590
    A(I,J)=WKAREA(JJ)                                  ZSC01600
    JJ=JJ+1                                           ZSC01610
30 CONTINUE                                             ZSC01620
IF (IER .EQ. 0) GO TO 9005                             ZSC01630
9000 CONTINUE                                          ZSC01640
CALL UERTST (IER,6HLEQT2F)                             ZSC01650
9005 RETURN                                           ZSC01660
END                                                     ZSC01670
```

```

SUBROUTINE LUDATF (A,LU,N,IA,IDGT,D1,D2,IPVT,EQUIL,WA,IER)      ZSC01680
DIMENSION             A(IA,1),LU(IA,1),IPVT(1),EQUIL(1)      ZSC01690
DOUBLE PRECISION     A,LU,D1,D2,EQUIL,WA,ZERO,ONE,FOUR,SIXTN,SIXTH, ZSC01700
*                   RN,WREL,BIGA,BIG,P,SUM,AI,WI,T,TEST,Q      ZSC01710
DATA                 ZERO,ONE,FOUR,SIXTN,SIXTH/0.DO,1.DO,4.DO,  ZSC01720
*                   16.DO,.0625DO/                             ZSC01730
IER = 0                                                       ZSC01740
RN = N                                                         ZSC01750
WREL = ZERO                                                    ZSC01760
D1 = ONE                                                       ZSC01770
D2 = ZERO                                                       ZSC01780
BIGA = ZERO                                                    ZSC01790
DO 10 I=1,N                                                    ZSC01800
    BIG = ZERO                                                  ZSC01810
    DO 5 J=1,N                                                  ZSC01820
        P = A(I,J)                                             ZSC01830
        LU(I,J) = P                                           ZSC01840
        P = DABS(P)                                           ZSC01850
        IF (P .GT. BIG) BIG = P                                ZSC01860
5    CONTINUE                                                  ZSC01870
    IF (BIG .GT. BIGA) BIGA = BIG                               ZSC01880
    IF (BIG .EQ. ZERO) GO TO 110                                ZSC01890
    EQUIL(I) = ONE/BIG                                         ZSC01900
10   CONTINUE                                                  ZSC01910
    DO 105 J=1,N                                               ZSC01920
        JM1 = J-1                                             ZSC01930
        IF (JM1 .LT. 1) GO TO 40                               ZSC01940
        DO 35 I=1,JM1                                         ZSC01950
            SUM = LU(I,J)                                       ZSC01960
            IM1 = I-1                                           ZSC01970
            IF (IDGT .EQ. 0) GO TO 25                           ZSC01980
            AI = DABS(SUM)                                       ZSC01990
            WI = ZERO                                           ZSC02000
            IF (IM1 .LT. 1) GO TO 20                             ZSC02010
            DO 15 K=1,IM1                                       ZSC02020
                T = LU(I,K)*LU(K,J)                               ZSC02030
                SUM = SUM-T                                       ZSC02040
                WI = WI+DABS(T)                                   ZSC02050
15        CONTINUE                                             ZSC02060
            LU(I,J) = SUM                                         ZSC02070
20        WI = WI+DABS(SUM)                                       ZSC02080
            IF (AI .EQ. ZERO) AI = BIGA                           ZSC02090
            TEST = WI/AI                                         ZSC02100
            IF (TEST .GT. WREL) WREL = TEST                       ZSC02110
            GO TO 35                                             ZSC02120
25        IF (IM1 .LT. 1) GO TO 35                               ZSC02130
            DO 30 K=1,IM1                                       ZSC02140
                SUM = SUM-LU(I,K)*LU(K,J)                       ZSC02150
30        CONTINUE                                             ZSC02160
            LU(I,J) = SUM                                         ZSC02170
35        CONTINUE                                             ZSC02180
40        P = ZERO                                               ZSC02190
            DO 70 I=J,N                                           ZSC02200
                SUM = LU(I,J)                                       ZSC02210
                IF (IDGT .EQ. 0) GO TO 55                         ZSC02220

```

```
AI = DABS(SUM) ZSC02230
WI = ZERO ZSC02240
IF (JM1 .LT. 1) GO TO 50 ZSC02250
DO 45 K=1, JM1 ZSC02260
  T = LU(I,K)*LU(K,J) ZSC02270
  SUM = SUM-T ZSC02280
  WI = WI+DABS(T) ZSC02290
45 CONTINUE ZSC02300
  LU(I,J) = SUM ZSC02310
50 WI = WI+DABS(SUM) ZSC02320
  IF (AI .EQ. ZERO) AI = BIGA ZSC02330
  TEST = WI/AI ZSC02340
  IF (TEST .GT. WREL) WREL = TEST ZSC02350
  GO TO 65 ZSC02360
55 IF (JM1 .LT. 1) GO TO 65 ZSC02370
  DO 60 K=1, JM1 ZSC02380
    SUM = SUM-LU(I,K)*LU(K,J) ZSC02390
60 CONTINUE ZSC02400
  LU(I,J) = SUM ZSC02410
65 Q = EQUIL(I)*DABS(SUM) ZSC02420
  IF (P .GE. Q) GO TO 70 ZSC02430
  P = Q ZSC02440
  IMAX = I ZSC02450
70 CONTINUE ZSC02460
  IF (RN+P .EQ. RN) GO TO 110 ZSC02470
  IF (J .EQ. IMAX) GO TO 80 ZSC02480
  D1 = -D1 ZSC02490
  DO 75 K=1, N ZSC02500
    P = LU(IMAX, K) ZSC02510
    LU(IMAX, K) = LU(J, K) ZSC02520
    LU(J, K) = P ZSC02530
75 CONTINUE ZSC02540
  EQUIL(IMAX) = EQUIL(J) ZSC02550
80 IPVT(J) = IMAX ZSC02560
  D1 = D1*LU(J, J) ZSC02570
85 IF (DABS(D1) .LE. ONE) GO TO 90 ZSC02580
  D1 = D1*SIXTH ZSC02590
  D2 = D2+FOUR ZSC02600
  GO TO 85 ZSC02610
90 IF (DABS(D1) .GE. SIXTH) GO TO 95 ZSC02620
  D1 = D1*SIXTN ZSC02630
  D2 = D2-FOUR ZSC02640
  GO TO 90 ZSC02650
95 CONTINUE ZSC02660
  JP1 = J+1 ZSC02670
  IF (JP1 .GT. N) GO TO 105 ZSC02680
  P = LU(J, J) ZSC02690
  DO 100 I=JP1, N ZSC02700
    LU(I, J) = LU(I, J)/P ZSC02710
100 CONTINUE ZSC02720
105 CONTINUE ZSC02730
  IF (IDGT .EQ. 0) GO TO 9005 ZSC02740
  P = 3*N+3 ZSC02750
  WA = P*WREL ZSC02760
  IF (WA+10.DO**(-IDGT) .NE. WA) GO TO 9005 ZSC02770
```

IER = 34	ZSC02780
GO TO 9000	ZSC02790
110 IER = 129	ZSC02800
D1 = ZERO	ZSC02810
D2 = ZERO	ZSC02820
9000 CONTINUE	ZSC02830
CALL UERTST(IER, 6HLUDATF)	ZSC02840
9005 RETURN	ZSC02850
END	ZSC02860
SUBROUTINE LUELMF (A,B,IPVT,N,IA,X)	ZSC02870
DIMENSION A(IA,1),B(1),IPVT(1),X(1)	ZSC02880
DOUBLE PRECISION A,B,X,SUM	ZSC02890
DO 5 I=1,N	ZSC02900
5 X(I) = B(I)	ZSC02910
IW = 0	ZSC02920
DO 20 I=1,N	ZSC02930
IP = IPVT(I)	ZSC02940
SUM = X(IP)	ZSC02950
X(IP) = X(I)	ZSC02960
IF (IW .EQ. 0) GO TO 15	ZSC02970
IM1 = I-1	ZSC02980
DO 10 J=IW,IM1	ZSC02990
SUM = SUM-A(I,J)*X(J)	ZSC03000
10 CONTINUE	ZSC03010
GO TO 20	ZSC03020
15 IF (SUM .NE. 0.DO) IW = I	ZSC03030
20 X(I) = SUM	ZSC03040
DO 30 IB=1,N	ZSC03050
I = N+1-IB	ZSC03060
IP1 = I+1	ZSC03070
SUM = X(I)	ZSC03080
IF (IP1 .GT. N) GO TO 30	ZSC03090
DO 25 J=IP1,N	ZSC03100
SUM = SUM-A(I,J)*X(J)	ZSC03110
25 CONTINUE	ZSC03120
30 X(I) = SUM/A(I,I)	ZSC03130
RETURN	ZSC03140
END	ZSC03150

```
SUBROUTINE LUREFF (A,B,UL,IPVT,N,IA,X,IDGT,RES,DX,IER)      ZSC03160
DIMENSION          A(IA,1),UL(IA,1),B(1),X(1),RES(1),DX(1),IPVT(1) ZSC03170
DIMENSION          ACCXT(2)                                  ZSC03180
DOUBLE PRECISION  A,ACCXT,B,UL,X,RES,DX,ZERO,XNORM,DXNORM   ZSC03190
DATA              ITMAX/75/,ZERO/0.DO/                      ZSC03200
IER=0                                                       ZSC03210
XNORM = ZERO                                               ZSC03220
DO 10 I=1,N                                                ZSC03230
    XNORM = DMAX1(XNORM,DABS(X(I)))                          ZSC03240
10 CONTINUE                                                ZSC03250
    IF (XNORM .NE. ZERO) GO TO 20                           ZSC03260
    IDGT = 50                                               ZSC03270
    GO TO 9005                                              ZSC03280
20 DO 45 ITER=1,ITMAX                                       ZSC03290
    DO 30 I=1,N                                             ZSC03300
    ACCXT(1) = 0.0D0                                       ZSC03310
    ACCXT(2) = 0.0D0                                       ZSC03320
        CALL VXADD(B(I),ACCXT)                               ZSC03330
        DO 25 J=1,N                                         ZSC03340
            CALL VXMUL(-A(I,J),X(J),ACCXT)                  ZSC03350
25 CONTINUE                                                ZSC03360
        CALL VXSTO(ACCXT,RES(I))                             ZSC03370
30 CONTINUE                                                ZSC03380
    CALL LUELMF (UL,RES,IPVT,N,IA,DX)                       ZSC03390
    DXNORM = ZERO                                           ZSC03400
    XNORM = ZERO                                           ZSC03410
    DO 35 I=1,N                                             ZSC03420
        X(I) = X(I) + DX(I)                                  ZSC03430
        DXNORM = DMAX1(DXNORM,DABS(DX(I)))                   ZSC03440
        XNORM = DMAX1(XNORM,DABS(X(I)))                       ZSC03450
35 CONTINUE                                                ZSC03460
    IF (ITER .NE. 1) GO TO 40                               ZSC03470
    IDGT = 50                                               ZSC03480
    IF (DXNORM .NE. ZERO) IDGT = -DLOG10(DXNORM/XNORM)     ZSC03490
40 IF (XNORM+DXNORM .EQ. XNORM) GO TO 9005                 ZSC03500
45 CONTINUE                                                ZSC03510
    IER = 129                                               ZSC03520
9000 CONTINUE                                              ZSC03530
    CALL UERTST(IER,6HLUREFF)                                ZSC03540
9005 RETURN                                                ZSC03550
    END                                                       ZSC03560

SUBROUTINE UERSET (LEVEL,LEVOLD)                            ZSC03570
INTEGER          LEVEL,LEVOLD                               ZSC03580
LEVOLD = LEVEL                                             ZSC03590
CALL UERTST (LEVOLD,6HUERSET)                              ZSC03600
RETURN                                                    ZSC03610
END                                                       ZSC03620
```

```

SUBROUTINE UERTST (IER,NAME)                                ZSC03630
INTEGER                                                    ZSC03640
   IEQ
INTEGER*2 NAME(3),NAMSET(3),NAMEQ(3)                       ZSC03650
DATA      NAMSET/2HUE,2HRS,2HET/                          ZSC03660
DATA      NAMEQ/2H ,2H ,2H /                              ZSC03670
DATA      LEVEL/4/,IEQDF/0/,IEQ/1H=/                      ZSC03680
IF (IER.GT.999) GO TO 25                                   ZSC03690
IF (IER.LT.-32) GO TO 55                                   ZSC03700
IF (IER.LE.128) GO TO 5                                    ZSC03710
IF (LEVEL.LT.1) GO TO 30                                  ZSC03720
CALL UGETIO(1,NIN,IOUNIT)                                 ZSC03730
IF (IEQDF.EQ.1) WRITE(IOUNIT,35) IER,NAMEQ,IEQ,NAME      ZSC03740
IF (IEQDF.EQ.0) WRITE(IOUNIT,35) IER,NAME                ZSC03750
GO TO 30.                                                 ZSC03760
5 IF (IER.LE.64) GO TO 10                                 ZSC03770
IF (LEVEL.LT.2) GO TO 30                                  ZSC03780
CALL UGETIO(1,NIN,IOUNIT)                                 ZSC03790
IF (IEQDF.EQ.1) WRITE(IOUNIT,40) IER,NAMEQ,IEQ,NAME      ZSC03800
IF (IEQDF.EQ.0) WRITE(IOUNIT,40) IER,NAME                ZSC03810
GO TO 30                                                  ZSC03820
10 IF (IER.LE.32) GO TO 15                                ZSC03830
IF (LEVEL.LT.3) GO TO 30                                  ZSC03840
CALL UGETIO(1,NIN,IOUNIT)                                 ZSC03850
IF (IEQDF.EQ.1) WRITE(IOUNIT,45) IER,NAMEQ,IEQ,NAME      ZSC03860
IF (IEQDF.EQ.0) WRITE(IOUNIT,45) IER,NAME                ZSC03870
GO TO 30                                                  ZSC03880
15 CONTINUE                                               ZSC03890
DO 20 I=1,3                                               ZSC03900
   IF (NAME(I).NE.NAMSET(I)) GO TO 25                     ZSC03910
20 CONTINUE                                               ZSC03920
   LEVOLD = LEVEL                                         ZSC03930
   LEVEL = IER                                            ZSC03940
   IER = LEVOLD                                           ZSC03950
   IF (LEVEL.LT.0) LEVEL = 4                              ZSC03960
   IF (LEVEL.GT.4) LEVEL = 4                              ZSC03970
   GO TO 30                                               ZSC03980
25 CONTINUE                                               ZSC03990
   IF (LEVEL.LT.4) GO TO 30                                ZSC04000
   CALL UGETIO(1,NIN,IOUNIT)                               ZSC04010
   IF (IEQDF.EQ.1) WRITE(IOUNIT,50) IER,NAMEQ,IEQ,NAME    ZSC04020
   IF (IEQDF.EQ.0) WRITE(IOUNIT,50) IER,NAME             ZSC04030
30 IEQDF = 0                                              ZSC04040
   RETURN                                                  ZSC04050
35 FORMAT(19H *** TERMINAL ERROR,10X,7H(IER = ,I3,      ZSC04060
1      20H) FROM IMSL ROUTINE ,3A2,A1,3A2)              ZSC04070
40 FORMAT(36H *** WARNING WITH FIX ERROR (IER = ,I3,    ZSC04080
1      20H) FROM IMSL ROUTINE ,3A2,A1,3A2)              ZSC04090
45 FORMAT(18H *** WARNING ERROR,11X,7H(IER = ,I3,      ZSC04100
1      20H) FROM IMSL ROUTINE ,3A2,A1,3A2)              ZSC04110
50 FORMAT(20H *** UNDEFINED ERROR,9X,7H(IER = ,I5,     ZSC04120
1      20H) FROM IMSL ROUTINE ,3A2,A1,3A2)              ZSC04130
55 IEQDF = 1                                              ZSC04140
   DO 60 I=1,3                                            ZSC04150
60 NAMEQ(I) = NAME(I)                                     ZSC04160
65 RETURN                                                  ZSC04170

```

END	ZSC04180
SUBROUTINE UGETIO(IOPT,NIN,NOUT)	ZSC04190
INTEGER IOPT,NIN,NOUT	ZSC04200
INTEGER NIND,NOUTD	ZSC04210
DATA NIND/5/,NOUTD/6/	ZSC04220
IF (IOPT.EQ.3) GO TO 10	ZSC04230
IF (IOPT.EQ.2) GO TO 5	ZSC04240
IF (IOPT.NE.1) GO TO 9005	ZSC04250
NIN = NIND	ZSC04260
NOUT = NOUTD	ZSC04270
GO TO 9005	ZSC04280
5 NIND = NIN	ZSC04290
GO TO 9005	ZSC04300
10 NOUTD = NOUT	ZSC04310
9005 RETURN	ZSC04320
END	ZSC04330
SUBROUTINE VXADD(A,ACC)	ZSC04340
DOUBLE PRECISION A,ACC(2)	ZSC04350
DOUBLE PRECISION X,Y,Z,ZZ	ZSC04360
X = ACC(1)	ZSC04370
Y = A	ZSC04380
IF (DABS(ACC(1)).GE.DABS(A)) GO TO 1	ZSC04390
X = A	ZSC04400
Y = ACC(1)	ZSC04410
1 Z = X+Y	ZSC04420
ZZ = (X-Z)+Y	ZSC04430
ZZ = ZZ+ACC(2)	ZSC04440
ACC(1) = Z+ZZ	ZSC04450
ACC(2) = (Z-ACC(1))+ZZ	ZSC04460
RETURN	ZSC04470
END	ZSC04480

SUBROUTINE VXMUL (A,B,ACC)	ZSC04490
DOUBLE PRECISION A,B,ACC(2)	ZSC04500
DOUBLE PRECISION X,HA,TA,HB,TB	ZSC04510
INTEGER IX(2),I	ZSC04520
LOGICAL*1 LX(8),LI(4)	ZSC04530
EQUIVALENCE (X,LX(1),IX(1)),(I,LI(1))	ZSC04540
DATA I/0/	ZSC04550
X = A	ZSC04560
LI(4) = LX(5)	ZSC04570
IX(2) = 0	ZSC04580
I = (I/16)*16	ZSC04590
LX(5) = LI(4)	ZSC04600
HA=X	ZSC04610
TA=A-HA	ZSC04620
X = B	ZSC04630
LI(4) = LX(5)	ZSC04640
IX(2) = 0	ZSC04650
I = (I/16)*16	ZSC04660
LX(5) = LI(4)	ZSC04670
HB = X	ZSC04680
TB = B-HB	ZSC04690
X = TA*TB	ZSC04700
CALL VXADD(X,ACC)	ZSC04710
X = HA*TB	ZSC04720
CALL VXADD(X,ACC)	ZSC04730
X = TA*HB	ZSC04740
CALL VXADD(X,ACC)	ZSC04750
X = HA*HB	ZSC04760
CALL VXADD(X,ACC)	ZSC04770
RETURN	ZSC04780
END	ZSC04790
SUBROUTINE VXSTO (ACC,D)	ZSC04800
DOUBLE PRECISION ACC(2),D	ZSC04810
D = ACC(1)+ACC(2)	ZSC04820
RETURN	ZSC04830
END	ZSC04840

SUBROUTINE ZSCNU	(X,N,FCN,NDIGIT,N1,A,Z,Y,XNORM,B,WK,	ZSC04850
1 MAXIT,PAR,IER)		ZSC04860
INTEGER	IER,MAXIT,N,NDIGIT,N1	ZSC04870
DOUBLE PRECISION	A(N1,1),B(1),PAR(1),WK(1),X(1),XNORM(1),	ZSC04880
1	Y(1),Z(N,1)	ZSC04890
INTEGER	I,IBMAX,IBNORM,IDGT,IEVAL,ITER,J,JER,JI,JS,	ZSC04900
1	MI,NRS,NSTART	ZSC04910
DOUBLE PRECISION	BIG,BNORM,CFACT,DX,EPS,HALF,HLMAX,RACC,REPS,	ZSC04920
1	RRX,RX,SFACT,SMALL,TEST,TN,TR	ZSC04930
DOUBLE PRECISION	DSEED	ZSC04940
DATA	SMALL/Z3410000000000000/	ZSC04950
IER = 0		ZSC04960
DSEED = 12345.0D0		ZSC04970
CFACT = 0.99D0		ZSC04980
BIG = 5.0D5		ZSC04990
SFACT = 0.1D0		ZSC05000
IBMAX = 50		ZSC05010
NRS = 2		ZSC05020
RACC = DMIN1(DMAX1(SMALL,10.0D0**(-NDIGIT)),0.1D0)		ZSC05030
REPS = DSQRT(SMALL)		ZSC05040
HLMAX = 3.0D0		ZSC05050
ITER = 0		ZSC05060
IEVAL = 0		ZSC05070
NSTART = 0		ZSC05080
IBNORM = 0		ZSC05090
RX = 1.0D0		ZSC05100
RRX = 0.0D0		ZSC05110
EPS = 0.0D0		ZSC05120
DO 5 I=1,N		ZSC05130
B(I) = 0.0D0		ZSC05140
A(N1,I) = 1.0D0		ZSC05150
Z(I,N1) = X(I)		ZSC05160
5 CONTINUE		ZSC05170
B(N1) = 1.0D0		ZSC05180
A(N1,N1) = 1.0D0		ZSC05190
JI = N1		ZSC05200
IEVAL = IEVAL+1		ZSC05210
CALL FCN (Z(1,N1),A(1,N1),N,PAR)		ZSC05220
BNORM = 0.0D0		ZSC05230
DO 10 I=1,N		ZSC05240
BNORM = BNORM+A(I,N1)*A(I,N1)		ZSC05250
10 CONTINUE		ZSC05260
XNORM(N1) = BNORM		ZSC05270
15 IF (NSTART.EQ.NRS) EPS = EPS*10.0D0		ZSC05280
IF (NSTART.EQ.0) EPS = DMIN1(RX,REPS)		ZSC05290
IF (EPS.GT.BIG) GO TO 120		ZSC05300
NSTART = MOD(NSTART,NRS)+1		ZSC05310
DO 30 J=1,N		ZSC05320
DO 25 I=1,N		ZSC05330
20 TR = (GGUBFS(DSEED)-0.5D0)*2.0D0		ZSC05340
IF (DABS(TR).LT.0.1D0) GO TO 20		ZSC05350
Z(I,J) = X(I)+DMAX1(DABS(X(I)),0.1D0)*TR*EPS		ZSC05360
25 CONTINUE		ZSC05370
IEVAL = IEVAL+1		ZSC05380
CALL FCN (Z(1,J),A(1,J),N,PAR)		ZSC05390

```
30 CONTINUE                                ZSC05400
   DO 35 J=1,N                              ZSC05410
       XNORM(J) = 0.DO                       ZSC05420
   DO 35 I=1,N                              ZSC05430
       XNORM(J) = XNORM(J)+A(I,J)*A(I,J)    ZSC05440
35 CONTINUE                                ZSC05450
40 JI = N1                                  ZSC05460
   JS = JI                                  ZSC05470
   DO 45 J=1,N                              ZSC05480
       IF (XNORM(J).GT.XNORM(JS)) JS = J    ZSC05490
       IF (XNORM(J).LT.XNORM(JI)) JI = J    ZSC05500
45 CONTINUE                                ZSC05510
   IF (XNORM(JI).EQ.0.DO) GO TO 125         ZSC05520
   IF (XNORM(JI).GT.SFACT*BNORM) GO TO 50  ZSC05530
   BNORM = XNORM(JI)                       ZSC05540
   IBNORM = ITER                           ZSC05550
50 IF ((ITER-IBNORM).GT.IBMAX) GO TO 120   ZSC05560
   ITER = ITER+1                           ZSC05570
   IF (ITER.GE.MAXIT) GO TO 115           ZSC05580
   DO 55 MI=1,N1                            ZSC05590
       Y(MI) = B(MI)                       ZSC05600
55 CONTINUE                                ZSC05610
   IDGT = 0                                ZSC05620
   CALL LEQT2F (A,1,N1,N1,Y,IDGT,WK,JER)  ZSC05630
   IF (JER.NE.0) GO TO 85                 ZSC05640
   DO 65 I=1,N                              ZSC05650
       DX = 0.ODO                           ZSC05660
       DO 60 J=1,N1                        ZSC05670
           DX = DX + Y(J)*Z(I,J)          ZSC05680
60 CONTINUE                                ZSC05690
   X(I) = DX                               ZSC05700
65 CONTINUE                                ZSC05710
   HALF = 0.DO                             ZSC05720
70 IEVAL = IEVAL+1                         ZSC05730
   CALL FCN (X,Y,N,PAR)                   ZSC05740
   TN = 0.DO                               ZSC05750
   DO 75 I=1,N                              ZSC05760
       TN = TN+Y(I)*Y(I)                  ZSC05770
75 CONTINUE                                ZSC05780
   IF (TN.LT.XNORM(JS)) GO TO 95          ZSC05790
   HALF = HALF+1.DO                       ZSC05800
   IF (HALF.GT.HLMAX) GO TO 85            ZSC05810
   DO 80 I=1,N                              ZSC05820
       X(I) = (X(I)+HALF*Z(I,JI))/(HALF+1.ODO) ZSC05830
80 CONTINUE                                ZSC05840
   GO TO 70                                ZSC05850
85 IF (JI.EQ.N1) GO TO 15                  ZSC05860
   XNORM(N1) = XNORM(JI)                  ZSC05870
   DO 90 I=1,N                              ZSC05880
       Z(I,N1) = Z(I,JI)                  ZSC05890
       A(I,N1) = A(I,JI)                  ZSC05900
90 CONTINUE                                ZSC05910
   GO TO 15                                ZSC05920
95 IF ((HALF.NE.0.DO).OR.(ITER.EQ.1)) GO TO 105 ZSC05930
   RX = SMALL                              ZSC05940
```

```
DO 100 I=1,N                                ZSC05950
    RX = DMAX1(RX,DABS(X(I)-Z(I,JI)))/DMAX1(DABS(X(I)),0.1D0)) ZSC05960
100 CONTINUE                                 ZSC05970
    RRX = DMAX1(-DLOG10(RX),0.0D0)           ZSC05980
    IF (RX.LE.RACC) GO TO 125               ZSC05990
105 IF (TN.LT.CFACT*KNORM(JI)) NSTART = 0  ZSC06000
    XNORM(JS) = TN                          ZSC06010
    DO 110 I=1,N                            ZSC06020
        Z(I,JS) = X(I)                     ZSC06030
        A(I,JS) = Y(I)                     ZSC06040
110 CONTINUE                                 ZSC06050
    GO TO 40                                ZSC06060
115 IER = 129                               ZSC06070
    GO TO 125                               ZSC06080
120 IER = 130                               ZSC06090
125 DO 130 I=1,N                            ZSC06100
    X(I) = Z(I,JI)                          ZSC06110
130 CONTINUE                                 ZSC06120
    RETURN                                   ZSC06130
    END                                     ZSC06140
```

APPENDIX C

RESULTS OF EXPERIMENTS - NEW SERIES

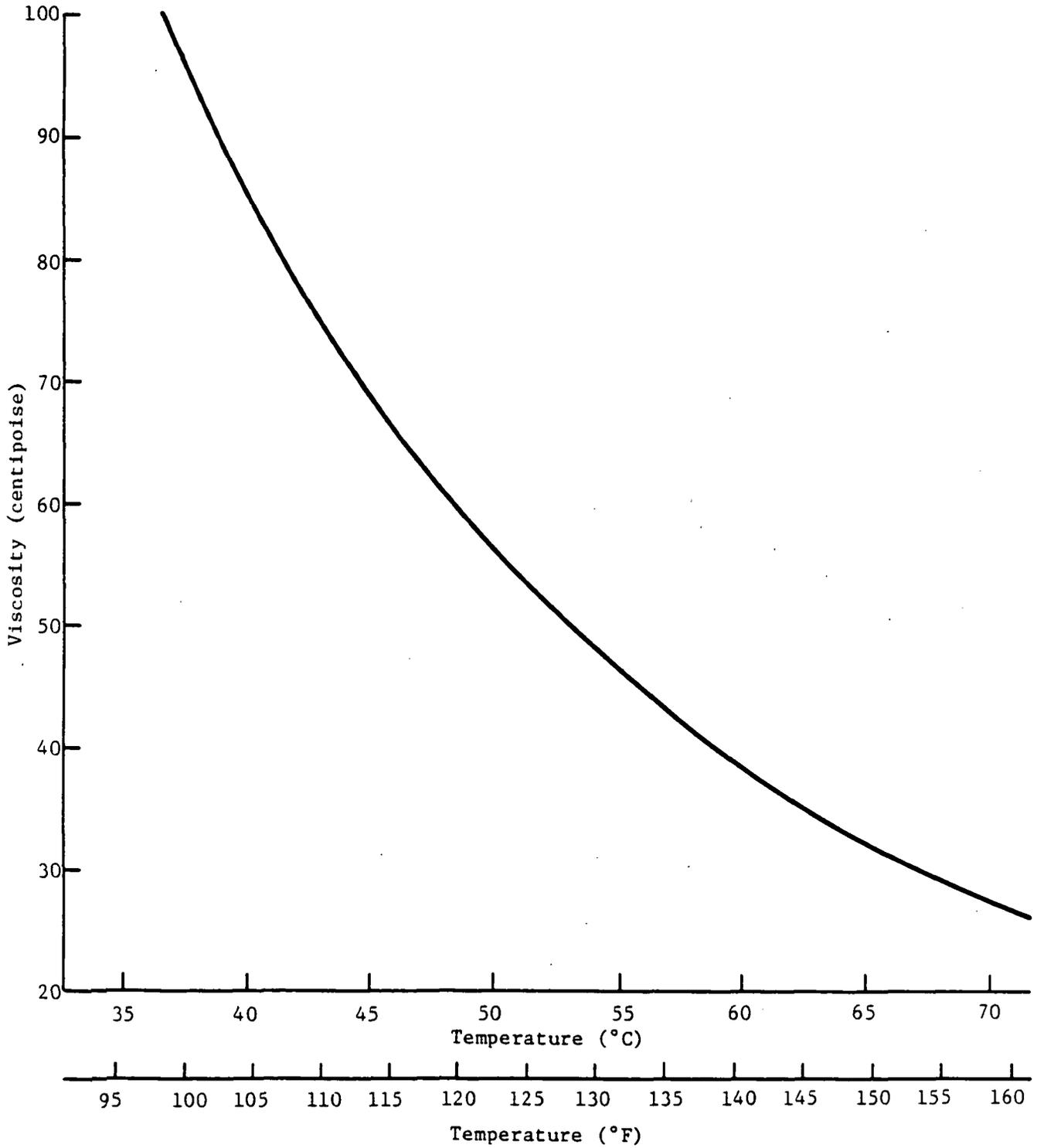


Fig. C-1 Viscosity of 20W40 Oil

TABLE C-1 PUMPING RING DESIGN A-1-A-1 - SMALL CLEARANCE; TiN BASED BABBIT SAE-11; STROKE = 2in

DATA POINT	FREQUENCY Hz	OIL OILF TEMPERATURE		CLAMP PRESSURE		PUMP PRESSURE		PUMP FLOW g/min
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)	
1	10	47.8	(118)	8.62	(1250)	8.87	(1287)	0
		47.2	(117)	↓	↓	7.31	(1060)	2.91
		47.8	(118)	↓	↓	4.79	(695)	3.40
		47.8	(118)	↓	↓	2.09	(303)	3.45
		47.8	(118)	↓	↓	0.04	(6)	3.67
2	35	50.0	(122)	8.62	(1250)	9.18	(1332)	0
		50.0	(122)	↓	↓	8.51	(1235)	15.6
		50.0	(122)	↓	↓	7.58	(1100)	16.2
		50.0	(122)	↓	↓	5.52	(800)	18.8
		48.9	(120)	↓	↓	2.76	(400)	20.4
		47.8	(118)	↓	↓	0.21	(30)	20.8
3	60	48.9	(120)	8.62	(1250)	9.10	(1320)	0
		49.4	(121)	↓	↓	8.81	(1278)	33.2
		50.0	(122)	↓	↓	6.23	(1005)	38.4
		48.9	(120)	↓	↓	5.52	(800)	40.3
		48.9	(120)	↓	↓	2.76	(400)	42.5
		49.4	(121)	↓	↓	0.33	(48)	43.0
4	10	47.8	(118)	6.89	(1000)	7.25	(1052)	0
		47.8	(118)	↓	↓	6.40	(929)	3.2
		47.8	(118)	↓	↓	5.19	(844)	3.5
		47.8	(118)	↓	↓	3.47	(504)	4.0
		47.8	(118)	↓	↓	-0-	(0)	4.3
5	35	47.8	(118)	6.89	(1000)	7.36	(1068)	0
		47.8	(118)	↓	↓	6.76	(980)	19.8
		47.8	(118)	↓	↓	5.52	(800)	21.5
		47.2	(117)	↓	↓	1.38	(200)	24.5
		46.7	(116)	↓	↓	0.12	(18)	14.9

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

STROKE = 2 W

TABLE C-1 CAST IRON PUMPING RING DESIGN - A-I-A-1 SMALL CLEARANCE MAT'L: TIO BASED BABBIT (SRE-11)

DATA POINT	FREQ. Hz	OIL INLET TEMP		CLAMPING PRESS		RUMPER PRESS		RUMPER FLOW g/min													
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)														
6	60	50.0	(122)	6.89	(1000)	7.39	(1072)	0													
		49.4	(121)			6.89	(1000)	42.0													
		49.4	(121)					4.82	(700)	46.8											
		49.4	(121)					2.75	(399)	49.4											
		50.0	(122)					0.14	(20)	50.1											
7	10	47.8	(118)	3.45	(500)	3.85	(558)	0													
		47.8	(118)			3.41	(495)	3.02													
		47.8	(118)			2.44	(354)	3.97													
		47.8	(118)			1.38	(200)	2.92													
		47.8	(118)			-0-	(0)	3.83													
8	35	48.3	(119)	3.45	(500)	3.86	(560)	0													
		48.9	(120)			3.35	(486)	20.9													
		50.0	(122)			2.37	(345)	22.7													
		49.4	(121)			1.38	(200)	25.6													
		50.0	(122)			0.10	(14)	28.0													
9	60	48.9	(120)	3.45	(500)	3.79	(550)	0													
		50.0	(122)			3.45	(500)	47.2													
		49.4	(121)			2.07	(300)	51.5													
		48.9	(120)			1.38	(200)	54.2													
		48.9	(120)			0.18	(26)	59.0													
10	10	47.8	(118)	8.62	(1250)	8.62	(1250)	2.42													
		47.8	(118)	6.89	(1000)	6.89	(1000)	3.15													
		47.2	(117)	5.52	(800)	5.52	(800)	3.05													
		47.2	(117)	4.31	(625)	4.31	(625)	2.00													
		47.8	(118)	1.33	(193)	1.33	(193)	0													

ORIGINAL FACE IS OF POOR QUALITY

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

183

TABLE C-1 CONT'D PUMPING RING DESIGN - A-1-A-1

MAT'L:

DATA POINT	FREQ. Hz	OIL INLET TEMP		Clamping Press		Ramped Press		Ramped Flow g/min
		°C	(°F)	MPa (lb/in ²)	MPa (lb/in ²)	MPa (lb/in ²)		
11	35	49.4	(121)	8.69	(1240)	8.69	(1240)	17.9
		49.4	(121)	6.99	(1014)	6.99	(1014)	21.7
		48.9	(120)	5.62	(816)	5.62	(816)	24.3
		50.0	(122)	4.27	(620)	4.27	(620)	23.2
		50.0	(122)	2.76	(400)	2.76	(400)	19.0
		50.0	(122)	1.38	(200)	1.38	(200)	19.0
		50.0	(122)	0.29	(42)	0.29	(42)	14.5
		50.0	(122)					
12	60	51.1	(124)	8.62	(1254)	8.62	(1254)	33.8
		50.0	(122)	6.87	(997)	6.87	(997)	41.0
		48.9	(120)	5.67	(822)	5.67	(822)	44.8
		48.9	(120)	4.22	(612)	4.22	(612)	44.8
		48.9	(120)	2.76	(400)	2.76	(400)	40.6
		48.9	(120)	1.40	(203)	1.40	(203)	36.8
		49.4	(121)	0.35	(51)	0.35	(51)	33.6
		49.4	(121)					
13	60	59.4	(139)	8.62	(1256)	8.69	(1261)	-0-
		58.9	(138)			7.95	(1080)	5.41
		60.6	(141)			4.96	(720)	6.43
		60.0	(140)			2.76	(400)	6.69
		60.0	(140)			1.08	(156)	6.74
14	60	47.8	(118)	8.62	(1250)	9.18	(1331)	-0-
		48.9	(120)			8.60	(1248)	4.83
		50.0	(122)			4.96	(720)	6.71
		48.9	(120)			2.76	(400)	7.26
		48.9	(120)			1.21	(176)	7.43

254mm (1.00in.) Stroke

254 (1.00in.) Stroke

181

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

TABLE C-2 PUMPING RUNG DESIGN - A-2-A-1 LARGE CLEARANCE, MAT'L: TIN BASED BABBITT (SAE-11)

DATA POINT	FREQ. Hz	OIL INLET TEMP		CLAMPING PRESS		RUMPER PRESS		Pumped Flow B/min
		°C	(°F)	MPa (lb/in ²)	MPa (lb/in ²)	MPa (lb/in ²)		
1	10	47.8	(118)	10.3	(1500)	10.5	(1517)	-0-
		47.8	(118)			8.64	(1254)	2.08
		47.8	(118)			7.57	(1099)	2.46
		47.8	(118)			4.82	(700)	2.85
		47.8	(118)			-0-	(0)	3.05
2	35	50.0	(122)	10.3	(1500)	10.8	(1572)	-0-
		50.0	(122)			8.62	(1250)	16.8
		48.3	(119)			7.78	(1128)	17.5
		47.8	(118)			4.14	(600)	18.6
		47.8	(118)			0.14	(20)	18.5
3	60	48.9	(120)	10.3	(1500)	10.8	(1575)	-0-
		48.9	(120)			10.2	(1488)	31.9
		50.0	(122)			9.06	(1315)	34.0
		50.0	(122)			5.88	(853)	35.6
		51.1	(124)			3.42	(496)	35.7
		50.0	(122)			2.44	(35)	35.9
4	10	48.9	(120)	8.62	(1250)	8.79	(1275)	-0-
		48.9	(120)			7.58	(1100)	2.93
		48.9	(120)			5.51	(800)	3.46
		48.9	(120)			2.67	(390)	3.64
		48.3	(119)			-0-	(0)	3.71
5	35	48.9	(120)	8.62	1250	8.98	(1302)	-0-
		48.9	(120)			8.35	(1224)	18.8
		48.9	(120)			6.76	(980)	20.2
		48.9	(120)			3.45	(500)	21.4
		48.3	(119)			0.23	(34)	21.8

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

185

TABLE C-2 CONT'D PUMPING RING DESIGN - A-2-A-1 LARGE CLEARANCE MAT'L: TiO BASED BARBITT (SAE-11)

DATA POINT	FREQ. Hz	OIL INLET TEMP		CLAMPING PRESS		PUMPED PRESS		PUMPED FLOW g/min
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)	
6	60	48.3	(119)	8.62	(1250)	9.13	(1324)	-0-
		49.4	(121)			8.77	(1272)	31.8
		50.0	(122)			7.80	(1131)	36.3
		50.0	(122)			6.02	(874)	39.7
		48.9	(120)			3.45	(500)	41.8
		49.4	(121)			0.14	(20)	41.9
		49.4	(121)					
7	10	50.0	(122)	6.89	(1000)	6.99	(1014)	-0-
		48.9	(120)			6.13	(884)	3.50
		48.9	(120)			4.81	(690)	4.32
		48.9	(120)			2.73	(396)	4.58
		48.9	(120)			0	-0-	4.69
8	35	47.8	(118)	6.89	(1000)	7.24	(1050)	-0-
		48.9	(120)			6.43	(933)	23.2
		48.9	(120)			6.33	(918)	20.8
		50.0	(122)			6.02	(873)	22.4
		48.3	(119)			3.45	(500)	23.5
		47.8	(118)			0.10	(14)	26.2
9	60	48.9	(120)	6.89	(1000)	7.25	(1052)	-0-
		48.9	(120)			7.20	(1044)	37.1
		48.9	(120)			6.09	(884)	44.2
		49.4	(121)			3.45	(500)	49.5
		48.9	(120)			0.15	(22)	50.6
10	10	48.9	(120)	10.3	(1500)	10.3	(1500)	1.70
		48.9	(120)			8.61	(1249)	2.84
		48.3	(119)			6.91	(1002)	4.25
		47.8	(118)			4.84	(702)	6.34
		48.3	(119)			2.76	(400)	5.16
		48.3	(119)			1.38	(200)	4.70
		47.8	(118)			0.10	(14)	7.96

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

TABLE C-2 CONT'D PUMPING RING DESIGN - A-2-A-1

MAT'L:

DATA POINT	REQ. Hz	OIL I.D. TEMP		CLAMPING PRESS		RUMPER PRESS		RUMPER FLOW g/min
		°C	(°F)	MPa (lb/in ²)	MPa (lb/in ²)	MPa (lb/in ²)		
11	35		(120)	10.3	(1500)	10.3	(1500)	19.2
			(118)	8.98	(1303)	8.98	(1303)	18.0
			(118)	7.58	(1100)	7.58	(1100)	20.9
			(117)	6.20	(900)	6.20	(900)	22.8
			(120)	4.85	(703)	4.85	(703)	22.4
			(121)	3.45	(500)	3.45	(500)	19.5
			(120)	2.07	(300)	2.07	(300)	18.8
			(120)	0.69	(100)	0.69	(100)	18.6
			(120)	0.23	(34)	0.23	(34)	18.3
12	60		(118)	10.3	(1500)	10.3	(1500)	34.1
			(119)	8.99	(1304)	8.99	(1304)	40.2
			(120)	7.58	(1099)	7.58	(1099)	45.2
			(121)	6.29	(912)	6.29	(912)	52.1
			(119)	4.84	(702)	4.84	(702)	55.6
			(119)	3.45	(500)	3.45	(500)	51.7
			(120)	2.07	(300)	2.07	(300)	46.0
			(120)	0.67	(97)	0.67	(97)	43.1
			(122)	0.42	(60)	0.42	(60)	40.7
13	60		(138)	8.62	(1250)	9.33	(1354)	0
			(137)	↓	↓	8.60	(1248)	5.71
			(137)	↓	↓	4.96	(720)	7.50
			(136)	↓	↓	2.76	(400)	7.60
			(138)	↓	↓	0.83	(120)	7.57

} 25.4 mm (1.000 in) STAINLESS

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

187

TABLE C-3 PUMPING RING DESIGN - A-1-B-1 SHORT LENGTH MAT'L: TIN BASED BARBITT (SAE-11)

DATA POINT	FREQ. Hz	OIL JOURNAL TEMP		CLAMPING PRESS		RUMPER PRESS		RUMPER FLOW g/min
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)	
1	10	47.8	(118)	10.3	(1500)	9.00	(1305)	-0-
		48.3	(119)	↓	↓	6.97	(982)	.659
		48.3	(119)	↓	↓	4.49	(651)	1.14
		47.3	(114)	↓	↓	2.23	(323)	1.62
		4.1	(15)	↓	↓	0.02	(4)	23.9
2	35	49.4	(121)	10.3	(1500)	9.82	(1424)	-0-
		50.0	(122)	↓	↓	7.35	(1072)	6.55
		50.0	(122)	↓	↓	4.91	(712)	10.3
		49.4	(121)	↓	↓	2.48	(360)	12.9
		49.4	(121)	↓	↓	0.04	(6)	15.1
3	60	50.6	(123)	10.3	(1500)	11.5	(1659)	-0-
		50.6	(123)	↓	↓	8.60	(1248)	22.9
		49.4	(121)	↓	↓	5.73	(831)	28.1
		48.9	(120)	↓	↓	2.91	(422)	32.1
		50.6	(123)	↓	↓	0.09	(13)	31.7
4	10	49.4	(121)	6.89	(1000)	6.95	(1008)	-0-
		50.0	(122)	↓	↓	5.20	(754)	1.42
		50.0	(122)	↓	↓	3.47	(503)	1.95
		50.0	(122)	↓	↓	1.74	(253)	2.30
		48.9	(120)	↓	↓	0.03	(5)	2.74
5	35	48.3	(119)	6.89	(1000)	7.69	(1116)	-0-
		47.2	(117)	↓	↓	5.76	(835)	9.08
		48.9	(120)	↓	↓	3.84	(557)	12.6
		49.4	(121)	↓	↓	1.93	(280)	14.8
		52.2	(126)	↓	↓	0.08	(11)	16.6

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES:

MTI

188

C-3

TABLE C-3 PUMPING RING DESIGN - A-1-B-1 SHORT LENGTH MAT'L: TIO BASED BARBITT (SAE-11)

DATA POINT	FREQ. Hz	OIL INLET TEMP		CLAMPING PRESS		RUMPER PRESS		Rumped Flow g/min
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)	
6	60	50.0	(122)	6.89	(1000)	7.69	(1116)	0
		50.0	(122)					22.6
		50.0	(122)					27.8
		50.6	(123)					32.0
		50.0	(122)					34.6
7	10	51.7	(125)	3.45	(500)	1.72	(250)	0
		51.1	(124)					.809
		51.7	(125)					2.02
		51.1	(124)					2.78
		52.2	(126)					3.24
8	35	51.1	(124)	3.45	(500)	3.52	(510)	-0-
		51.1	(124)					5.4
		51.7	(125)					12.4
		52.2	(126)					14.6
		51.1	(124)					18.6
9	60	52.8	(127)	3.45	(500)	3.71	(538)	-0-
		50.6	(123)					190
		51.1	(124)					259
		52.2	(126)					31.0
		53.3	(128)					36.1
10	10	47.2	(117)	10.3	(1500)	10.3	(1500)	-0-
		47.2	(117)					.693
		47.2	(117)					1.24
11	35	47.2	(117)	10.3	(1500)	10.3	(1500)	6.08
		48.9	(120)					9.40
		48.3	(119)					12.8

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTJ

189

ORIGINAL PAGE IS OF POOR QUALITY

TABLE C-3 PUMPING RING DESIGN - A-B-1 SHORT LENGTH MAT'L:

DATA POINT	TRIP. Hz	OIL INLET TEMP		CLAMPING PRESS		PUMPED PRESS		PUMPED FLOW g/min											
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)												
12	60	48.9	(120)	10.3	(1500)	10.3	(1500)	10.5											
		49.4	(121)	8.62	(1250)	8.62	(1250)	30.1											
		50.0	(122)	6.89	(1000)	6.89	(1000)	38.8											
13	60	58.3	(137)	8.62	(1250)	7.90	(1144)	-0-											
		58.9	(138)	↓	↓	7.03	(1020)	1.6											
		58.3	(137)			4.96	(720)	4.68	} 25.4 mm (1.00 in) Stroke										
		57.7	(136)			2.76	(400)	5.38											
		58.3	(137)	↓	↓	0.59	(86)	5.55											
14	0	51.1	(124)	0	-0-	0	-0-	1.42											
	10	51.1	(124)	0	-0-	0	-0-	2.34											
	35	51.1	(124)	0	-0-	0	-0-	12.1 (6.9)										Inlet P = 0.076 MPa (11 lb/in ²)	
	60	51.1	(124)	0	-0-	0	-0-	17.8											
15	0	53.3	(128)	0	-0-	0	-0-	4.6											
	10	53.3	(128)	0	-0-	0	-0-	9.74											
	35	53.3	(128)	0	-0-	0	-0-	11.7										Inlet P = 0.274 MPa (40 lb/in ²)	
	60	53.3	(128)	0	-0-	0	-0-	22.4											
16	0	50.6	(123)	0	-0-	0	-0-	2.00											
	10	50.6	(123)	0	-0-	0	-0-	5.99											
	35	50.0	(122)	0	-0-	0	-0-	9.88											
	60	51.7	(125)	0	-0-	0	-0-	20.7										Inlet P = 0.178 MPa (26 lb/in ²)	
17	0	50.0	(122)	6.89	(1000)	0	-0-	-											
	10	50.0	(122)	↓	↓	0	-0-	2.69											
	35	50.0	(122)			0	-0-	15.0											
	60	50.6	(123)	↓	↓	0	-0-	43.4										Inlet P = 0.138 MPa (20 lb/in ²)	

061

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

TABLE C-4. PUMPING RING DESIGN - A-1-A-2 CONTOURED MAT'L: T10 BASED BABBITT (SAE-11)

DATA POINT	FREQ. Hz	OIL INLET TEMP		CLAMPING PRESS		RUMPER PRESS		RUMPER RAW g/min
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)	
1	10	47.8	(118)	8.62	(1250)	8.98	(1303)	-0-
		48.3	(119)			8.62	(1250)	2.64
		47.8	(118)			4.96	(720)	2.52
		47.8	(118)			2.76	(400)	2.40
		47.8	(118)			0.43	(60)	2.69
2	35	50.6	(123)	8.62	(1250)	9.45	(1372)	-0-
		50.0	(122)			8.62	(1250)	18.0
		47.2	(117)			4.96	(720)	17.2
		48.3	(119)			2.76	(400)	17.3
		48.9	(120)			0.78	(114)	17.2
3	60	50.6	(123)	8.62	1250	9.51	(1380)	-0-
		48.9	(120)			8.62	(1250)	39.0
		47.2	(117)			6.70	(972)	38.0
		47.2	(117)			2.71	(393)	36.8
		49.4	(121)			1.03	(150)	33.0
4	10	47.2	(117)	6.89	(1000)	7.16	(1039)	-0-
		48.3	(119)			6.54	(944)	1.30
		48.3	(119)			4.96	(720)	1.40
		48.3	(119)			2.76	(400)	1.56
		48.9	(120)			0.29	(42)	1.60
5	35	49.4	(121)	6.89	(1000)	7.51	(1090)	-0-
		48.9	(120)			6.89	(1000)	18.9
		49.4	(121)			4.96	(720)	18.6
		48.9	(120)			2.55	(370)	18.8
		48.3	(119)			1.36	(198)	18.9

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

TABLE C-4 PUMPING RING DESIGN - A-1-A-2 CONTROLLED MAT'L: T3 BARS BABBIT (SAE-11)

DATA POINT	FREQ. Hz	OIL INLET TEMP		CLAMPING PRESS		PUMPING PRESS		PUMPED FLOW g/min
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)	
6	60	50.6	(123)	↓	↓	7.58	(1092)	-0-
		50.0	(122)			6.70	(972)	47.2
		48.9	(120)			5.09	(738)	48.2
		47.2	(117)			2.76	(400)	46.8
		50.6	(123)			1.74	(252)	45.2
7	10	49.4	(121)	↓	↓	3.96	(574)	-0-
		50.0	(122)			2.90	(420)	4.02
		50.0	(122)			1.65	(240)	4.00
		50.0	(122)			0.76	(53)	4.00
8	35	47.8	(118)	↓	↓	4.09	(594)	-0-
		48.9	(120)			3.31	(480)	36.5
		50.0	(122)			2.07	(300)	38.2
		50.0	(122)			1.65	(240)	38.4
9	60	48.9	(120)	↓	↓	4.30	(624)	-0-
		48.9	(120)			3.31	(480)	80.8
10	10	47.2	(117)	↓	↓	9.10	(1320)	0.456
		47.8	(118)			6.89	(1000)	1.74
		47.8	(118)			5.51	(800)	2.78
		47.2	(117)			4.14	(600)	4.20
		47.8	(118)			2.76	(400)	6.00
11	35	47.8	(118)	↓	↓	9.10	(1320)	15.4
		49.4	(121)			6.89	(1000)	26.3
		49.4	(121)			5.51	(800)	26.8
		47.8	(118)			4.14	(600)	33.2
		47.2	(117)			2.76	(400)	41.2

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

TABLE C-4 PUMPING RING DESIGN - A-1-A-2 CONTOURED MAT'L: TiO BASED BACONIT (SAE-11)

DATA POINT	FRIQ. Hz	OIL INLET TEMP		CLAMPING PRESS		PUMPED PRESS		PUMPED FLOW g/min											
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)												
12	60	48.9	(120)	9.10	(1320)	9.10	(1320)	31.2											
		49.4	(121)	7.02	(1018)	7.02	(1018)	46.4											
		50.0	(122)	5.58	(810)	5.58	(810)	59.2											
		48.9	(120)	4.14	(600)	4.14	(600)	74.0											
		48.9	(120)	3.07	(446)	3.07	(446)	95.8											
13	60	57.7	(136)	8.62	(1250)	9.20	(1335)	-0-	} 25.4mm (1.00in) STAINL										
		58.3	(137)	↓	↓	8.60	(1248)	2.31											
		58.3	(137)			4.96	(720)	2.72											
		58.9	(138)			2.76	(400)	3.85											
		57.7	(136)			0.52	(76)	4.29											

194

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

TABLE C-5 PUMPING RING DESIGN - B-1-A-1

MAT'L: Rulon J

DATA POINT	FREQ. Hz	OIL JCT. TEMP		CLAMPING PRESS		PUMPING PRESS		PUMPED FLOW g/min
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)	
1	10	48.3	(119)	5.17	(750)	4.92	(714)	-0-
		47.8	(118)			3.31	(480)	4.08
		48.9	(120)			1.65	(240)	4.48
		48.3	(119)			0.83	(120)	4.40
		48.3	(119)			0.13	(19)	4.30
2	35	47.8	(118)	5.17	(750)	5.20	(755)	-0-
		47.2	(117)			3.31	(480)	25.7
		49.4	(121)			1.65	(240)	28.2
		48.3	(119)			0.83	(120)	27.8
		47.8	(118)			0.26	(37)	27.0
3	60	48.9	(120)	5.17	(750)	5.18	(751)	-0-
		48.9	(120)			5.38	(480)	59.9
		48.9	(120)			1.65	(240)	65.4
		48.3	(119)			0.83	(120)	63.0
		48.3	(119)			0.63	(91)	61.7
4	10	47.2	(117)	3.45	(500)	3.31	(578)	-0-
		47.8	(118)			2.48	(360)	3.63
		48.9	(120)			1.65	(240)	5.77
		48.9	(120)			0.83	(120)	5.48
		48.3	(119)			0.14	(20)	6.05
5	35	48.3	(119)	3.45	(500)	3.57	(518)	-0-
		48.3	(119)			2.48	(360)	32.5
		48.3	(119)			1.65	(240)	41.6
		48.9	(120)			0.83	(120)	39.7
		48.3	(119)			0.32	(47)	39.7

CHARGE NO.

PROJECT TITLE:

ENGINEER:

DATE:

NOTES:

MTI

TABLE C-5 PUMPING RING DESIGN - B-1-A-1

MAT'L: RUBON J

DATA POINT	FREQ. Hz	OIL I.D. - T. TEMP		Demand Press		Ramped Press		Ramped Flow g/min			
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)				
6	60	48.9	(120)	3.45	(500)	3.52	(510)	-0-			
		48.9	(120)	↓	↓	2.48	(360)	77.1			
		49.4	(121)	↓	↓	1.65	(240)	101			
		48.9	(120)	↓	↓	0.83	(120)	113			
		50.0	(122)	↓	↓	0.51	(74)	89.2			
7	10	48.9	(120)	1.72	(250)	1.98	(288)	-0-			
		48.9	(120)	↓	↓	1.65	(240)	6.38			
		48.3	(119)	↓	↓	1.24	(180)	7.37			
		48.3	(119)	↓	↓	0.83	(120)	7.87			
		48.9	(120)	↓	↓	0.30	(43)	11.5			
8	35	49.4	(121)	1.72	(250)	2.08	(301)	-0-			
		50.6	(123)	↓	↓	1.65	(240)	48.6			
		49.4	(121)	↓	↓	1.24	(180)	53.7			
		49.4	(121)	↓	↓	1.00	(146)	61.8			
9	60	50.0	(122)	1.72	(250)	2.05	(295)	-0-			
		48.3	(119)	↓	↓	1.65	(240)	95.4			
		49.4	(121)	↓	↓	1.24	(180)	140			
		48.9	(120)	↓	↓	1.13	(164)	136			
10	10	48.3	(119)	4.14	(600)	4.14	(600)	-0-			
		49.4	(121)	3.45	(500)	3.45	(500)	1.24			
		48.3	(119)	1.72	(250)	1.72	(250)	6.49			
11	35	48.9	(120)	4.14	(600)	4.14	(600)	-0-			
		48.9	(120)	3.45	(500)	3.45	(500)	8.97			
		48.9	(120)	1.72	(250)	1.72	(250)	43.2			
12	60	47.8	(118)	3.79	(550)	3.79	(550)	-0-			
		49.4	(121)	3.45	(500)	3.45	(500)	16.5			
		47.8	(118)	1.72	(250)	1.72	(250)	105			
CHARGE NO.		PROJECT TITLE:			ENGINEER:		DATE:		NOTES:		MTI

961

TABLE C-5 PUMPING RING DESIGN - B-1-A-1 MAT'L: RULAN J

DATA POINT	FRIG. Hz	OIL INLET TEMP		CLAMPING PRESS		PUMPED PRESS		Pumped Flow g/min
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)	
13	60	58.3	(131)	5.17	(756)		(847)	-0-
		58.9	(138)	↓	↓		(800)	9.72
		59.4	(139)				(600)	11.2
		58.9	(138)				(396)	10.8
		58.3	(131)				(118)	10.1
14	35	48.3	119	0	-0-	0.58	72	8.37
		48.3	119	0.34	50	0.58	85	10.9
		49.4	121	0.69	100	0.87	126	18.5
		49.4	121	1.03	150	0.90	130	58.6
		48.9	120	1.38	200	0.98	142	67.8
		48.3	119	1.72	250	1.11	161	66.6
		48.3	119	2.07	300	1.06	154	73.8
		48.3	119	2.41	350	0.78	119	87.6
		48.9	120	2.76	400	0.68	99	64.7
		48.9	120	3.10	450	0.63	91	43.7
		49.4	121	1.72	250	1.17	170	62.1
48.9	120	1.38	200	1.23	179	63.6		
15	35	47.8	118	0	-0-	0	-0-	90.9
		48.3	119	0.34	50	↓	↓	76.3
		48.9	120	0.69	100			87.8
		48.9	120	1.03	150			87.7
		48.9	120	1.38	200			78.9
		48.3	119	1.72	250			82.4
		48.3	119	2.07	300			86.0
		48.9	120	2.41	350	↓	↓	53.7
		48.9	120	2.76	400			49.4

197

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

TABLE C-6 PUMPING RING DESIGN - C-1-A-1

MAT'L: OREGON GRAPHITE (CNF-T)

DATA POINT	TRIP. Hz	OIL INLET TEMP		CLAMPING PRESS		PUMPED PRESS		Pumped Flow g/min
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)	
1	10	48.9	(120)	6.89	(1000)	7.04	(1022)	-0-
		41.4	(121)			6.62	(961)	1.51
		48.9	(120)			5.50	(798)	4.27
		49.4	(121)			2.76	(400)	5.58
		48.9	(120)			1.36	(198)	5.20
		48.9	(120)			0.37	(54)	5.0
2	35	50.6	(123)	6.89	(1000)	7.35	(1066)	-0-
		51.1	(124)			6.89	(1000)	8.16
		48.9	(120)			5.50	(798)	13.14
		49.4	(121)			2.76	(400)	29.9
		48.9	(120)			1.36	(198)	29.3
		48.9	(120)			0.23	(34)	26.2
3	60	47.8	(118)	6.89	(1000)	7.44	(1080)	-0-
		47.2	(117)			6.89	(1000)	30.2
		48.3	(119)			5.50	(798)	44.6
		49.4	(121)			2.76	(400)	57.0
		49.4	(121)			1.36	(198)	55.7
		49.4	(121)			0.50	(72)	53.4
4	10	47.2	(117)	5.17	(750)	3.98	(578)	-0-
		47.8	(118)			3.72	(540)	4.95
		48.9	(120)			2.48	(360)	6.92
		48.9	(120)			1.24	(180)	6.77
		48.9	(120)			0.72	(102)	6.38
5	35	48.9	(120)	5.17	(750)	5.09	(738)	-0-
		50.0	(122)			3.72	(540)	27.8
		48.9	(120)			2.07	(300)	35.6
		48.9	(120)			1.24	(180)	36.0
		50.0	(122)			0.35	(51)	35.6

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

198

TABLE C-6 PUMPING RING DESIGN - C-1-A-1

MAT'L: CARBON GRAPHITE (CNF-J)

DATA POINT	FREQ. Hz	OIL INLET TEMP		CLAMPING PRESS		PUMPED PRESS		PUMPED FLOW g/min
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)	
6	60	50.6	(123)	5.17	(750)	5.41	(785)	-0-
		50.6	(123)			3.72	(540)	54.6
		49.4	(121)			2.07	(300)	68.0
		49.4	(121)			1.24	(180)	71.2
		47.8	(118)			0.47	(68)	71.6
7	10	48.3	(119)	3.45	(500)	2.15	(312)	-0-
		48.3	(119)			1.65	(240)	4.11
		48.3	(119)			1.24	(180)	7.00
		48.9	(120)			0.83	(120)	8.08
		48.3	(119)			0.16	(24)	9.23
8	35	49.4	(121)	3.45	(500)	2.96	(429)	-0-
		49.4	(121)			2.48	(360)	13.4
		49.4	(121)			1.65	(240)	26.8
		48.9	(120)			0.83	(120)	36.0
		50.0	(122)			0.41	(60)	38.4
9	60	47.2	(117)	3.45	(500)	3.41	(495)	-0-
		47.2	(117)			2.48	(360)	35.2
		47.8	(118)			1.65	(240)	64.0
		47.8	(118)			0.96	(140)	64.0
10	10	WILL NOT SELF SUSTAIN						
11	35	49.4	(121)	7.38	(1070)	7.38	(1070)	10.2
		50.0	(122)	6.89	(1000)	6.89	(1000)	10.6
		WILL NOT SELF SUSTAIN AT LOWER PRESSURES						
12	60	51.1	(124)	7.73	(1121)	7.73	(1121)	24.6
		49.4	(121)	6.89	(1000)	6.89	(1000)	26.9
		50.0	(122)	6.20	(900)	6.20	(900)	27.5
WILL NOT SELF SUSTAIN AT LOWER PRESSURES								

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

199

TABLE C-7 PUMPING RING DESIGN D-1-A-1 (12mm BORE) MAT'L: TiO BASED RABBIT (SAE-11)

DATA POINT	FREQ. Hz	OIL JOURNAL TEMP		CLAMPING PRESS		PUMPING PRESS		PUMPED FLOW g/min
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)	
1	10	48.3	(119)	8.62	(1250)	9.02	(1308)	-0-
		49.4	(121)			7.44	(1080)	1.88
		48.3	(119)			4.93	(720)	1.82
		48.3	(119)			3.10	(450)	1.75
		48.3	(119)			0.12	(17)	1.77
2	35	49.4	(121)	8.62	(1250)	9.38	(1341)	-0-
		50.0	(122)			7.44	(1080)	14.6
		50.0	(122)			4.93	(720)	14.4
		48.3	(119)			3.10	(450)	14.2
		48.3	(119)			0.97	(41)	13.4
3	60	49.4	(121)	8.62	(1250)	9.31	(1350)	-0-
		48.3	(119)			7.44	(1080)	31.3
		48.3	(119)			4.93	(720)	29.7
		49.4	(121)			3.10	(450)	27.6
		49.4	(121)			0.21	(31)	25.7
4	10	48.9	(120)	6.89	(1000)	7.43	1077	-0-
		48.9	(120)			5.38	780	2.65
		48.3	(119)			4.14	600	2.60
		47.8	(118)			1.65	240	2.40
		49.4	(121)			0.12	17	2.30
5	35	47.8	(118)	6.89	(1000)	7.51	(1090)	-0-
		48.3	(119)			6.62	(960)	17.1
		50.0	(122)			5.17	(750)	16.6
		48.3	(119)			3.10	(450)	16.6
		48.3	(119)			0.28	(41)	15.8

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

200

TABLE C-7 PUMPING RING DESIGN - D1-A-1 (12mm BORE) MAT'L: TIJ BASED BABBITT (SAE-11)

DATA POINT	FREQ. Hz	OIL INLET TEMP		CLAMPING PRESS		PUMPED PRESS		PUMPED FLOW g/min		
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)			
6	60	47.8	(118)	6.89	(1000)	7.49	(1086)	-0-		
		47.8	(118)					6.20	(900)	40.2
		50.0	(122)					4.14	(600)	37.7
		49.4	(121)					1.65	(240)	35.1
		49.4	(121)					0.41	(59)	33.1
7	10	48.3	(119)	3.45	(500)	3.72	(540)	-0-		
		48.9	(120)					2.89	(420)	2.88
		48.3	(119)					1.65	(240)	8.67
		48.3	(119)					0.12	(18)	4.20
8	35	44.4	(121)	3.45	(500)	4.01	(582)	-0-		
		49.4	(121)					2.90	(420)	20.8
		49.4	(121)					1.65	(240)	21.7
		44.4	(121)					0.30	(43)	22.2
9	60	50.0	(122)	3.45	(500)	4.01	(582)	-0-		
		47.8	(118)					2.90	(420)	39.6
		47.8	(118)					1.65	(240)	43.0
		50.0	(122)					0.32	(47)	44.2
10	10	45.6	(114)	8.62	(1250)	8.62	(1250)	1.35		
		47.2	(117)					6.89	(1000)	1.74
		48.3	(119)					5.51	(800)	2.23
		48.9	(120)					4.14	(600)	2.76
		48.9	(119)					2.76	(400)	3.54
		48.3	(119)					1.39	(200)	3.80

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

201

TABLE C-1 PUMPING RING DESIGN -- D-1-A-1 (12mm 802) MATL: TIN BASED BABBIT (SAE 11)

DATA POINT	FREQ Hz	OIL INLET TEMP		CLAMPING PRESS		PUMPED PRESSURE		PUMPED FLOW ml/d											
		°C	(°F)	MPa	(lb/in ²)	MPa	(lb/in ²)												
11	35	48.3	(119)	8.62	(1250)	8.62	(1250)	13.0											
		48.5	(119)	6.89	(1000)	6.89	(1000)	15.4											
		49.4	(121)	5.51	(800)	5.51	(800)	17.6											
		48.9	(120)	4.14	(600)	4.14	(600)	20.0											
		49.4	(121)	2.76	(400)	2.76	(400)	21.0											
		48.4	(120)	1.39	(200)	1.39	(200)	19.3											
12	60	48.3	(119)	8.62	(1250)	8.62	(1250)	22.3											
		48.9	(120)	6.89	(1000)	6.89	(1000)	26.1											
		47.8	(118)	5.51	(800)	5.51	(800)	40.6											
		47.8	(118)	4.14	(600)	4.14	(600)	43.2											
		47.8	(118)	2.76	(400)	2.76	(400)	40.1											
		47.8	(118)	1.39	(200)	1.39	(200)	37.2											
13	60	48.3	(119)	8.62	(1250)	9.75	(1415)	0											
		47.8	(118)	↓	↓	8.27	(1200)	10.4											
		48.3	(119)	↓	↓	6.62	(960)	10.1											
		50.6	(123)	↓	↓	4.14	(600)	9.8											
		48.3	(119)	↓	↓	2.48	(360)	9.0											
		49.4	(121)	↓	↓	0.17	(25)	8.7											
14	60	58.9	(138)	8.62	(1250)	9.55	(1386)	-0-											
		59.4	(139)	↓	↓	8.27	(1200)	7.71											
		60.0	(140)	↓	↓	6.62	(960)	7.56											
		59.4	(139)	↓	↓	4.14	(600)	7.20											
		59.4	(139)	↓	↓	2.48	(360)	6.73											
		60.0	(140)	↓	↓	0.12	(17)	6.43											

25.4 mm (1.0012) Stroke

25.4 mm (1.0012) Stroke

CHARGE NO. PROJECT TITLE: ENGINEER: DATE: NOTES: MTI

202

TABLE D-1

PARAMETERS USED IN CALCULATIONS

RUN	MATERIAL	R mm (in.)	$C \cdot 10^3$ mm (in.)	t mm (in.)	E MPa $\cdot 10^3$ (psi $\cdot 10^6$)	ν	L_1 mm (in.)	L mm (in.)	e mm (in.)
I	Babbitt	9.52 (0.375)	11.43 (0.45)	1.19 (.047)	51.7 (7.5)	0.36	7.56 (0.298)	6.78 (0.267)	2.92 (.115)
II	Babbitt	9.52 (0.375)	19.05 (0.45)	1.19 (.047)	51.7 (7.5)	0.36	7.56 (0.298)	6.78 (0.267)	2.92 (.115)
V	Rulon J	9.52 (0.375)	42.55 (1.675)	2.44 (.096)	1.72 (.25)	0.46	7.56 (0.298)	6.78 (0.267)	2.92 (.115)
VI	Carbon	9.52 (0.375)	20.35 (0.80)	1.44 (.056)	21.5 (3.11)	0.29	7.56 (0.298)	6.78 (0.267)	2.92 (.115)
VII	Babbitt	6.00 (0.236)	8.89 (0.35)	0.89 (.035)	51.7 (7.5)	0.36	5.59 (0.22)	5.08 (0.2)	1.93 (.076)

*The viscosity used is that corresponding to test temperature plus an assumed 20°F used in the oil film.

FIGURES SHOWING COMPARISON WITH EXPERIMENTAL DATA
AND UNCORRECTED THEORY

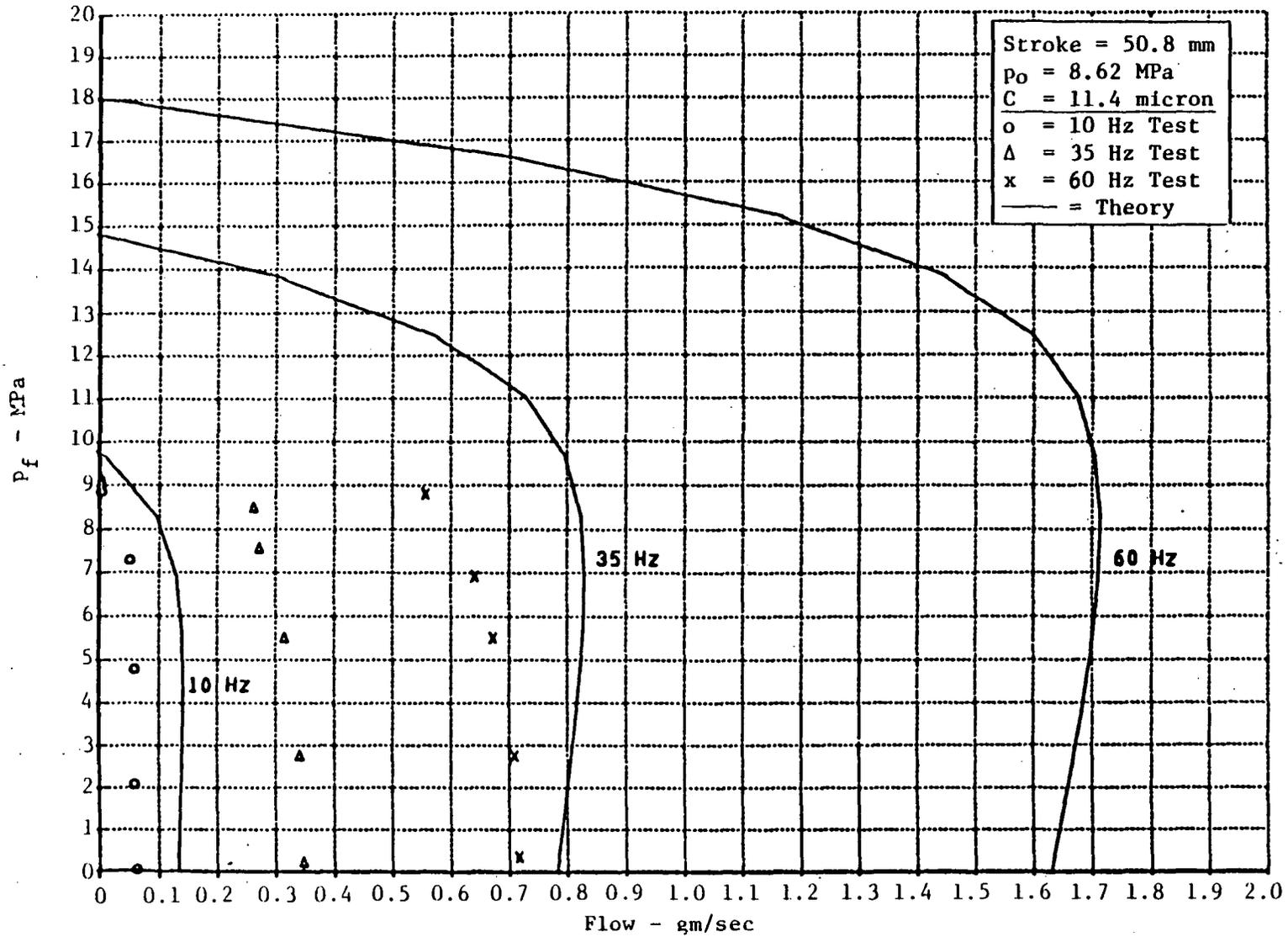


Fig. D-1 Performance of Babbitt Pumping Ring (I)

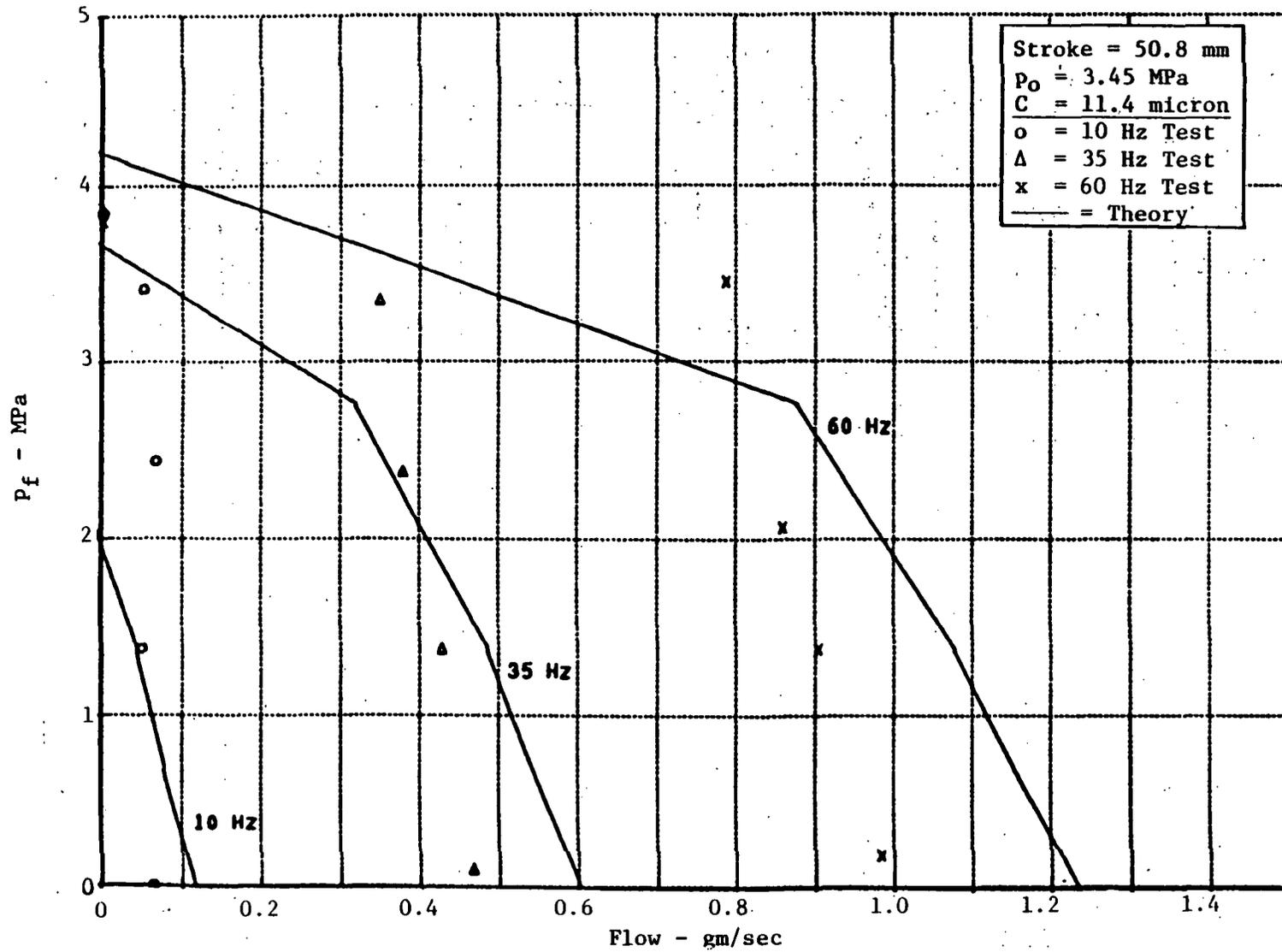


Fig. D-3 Performance of Babbitt Pumping Ring (I)

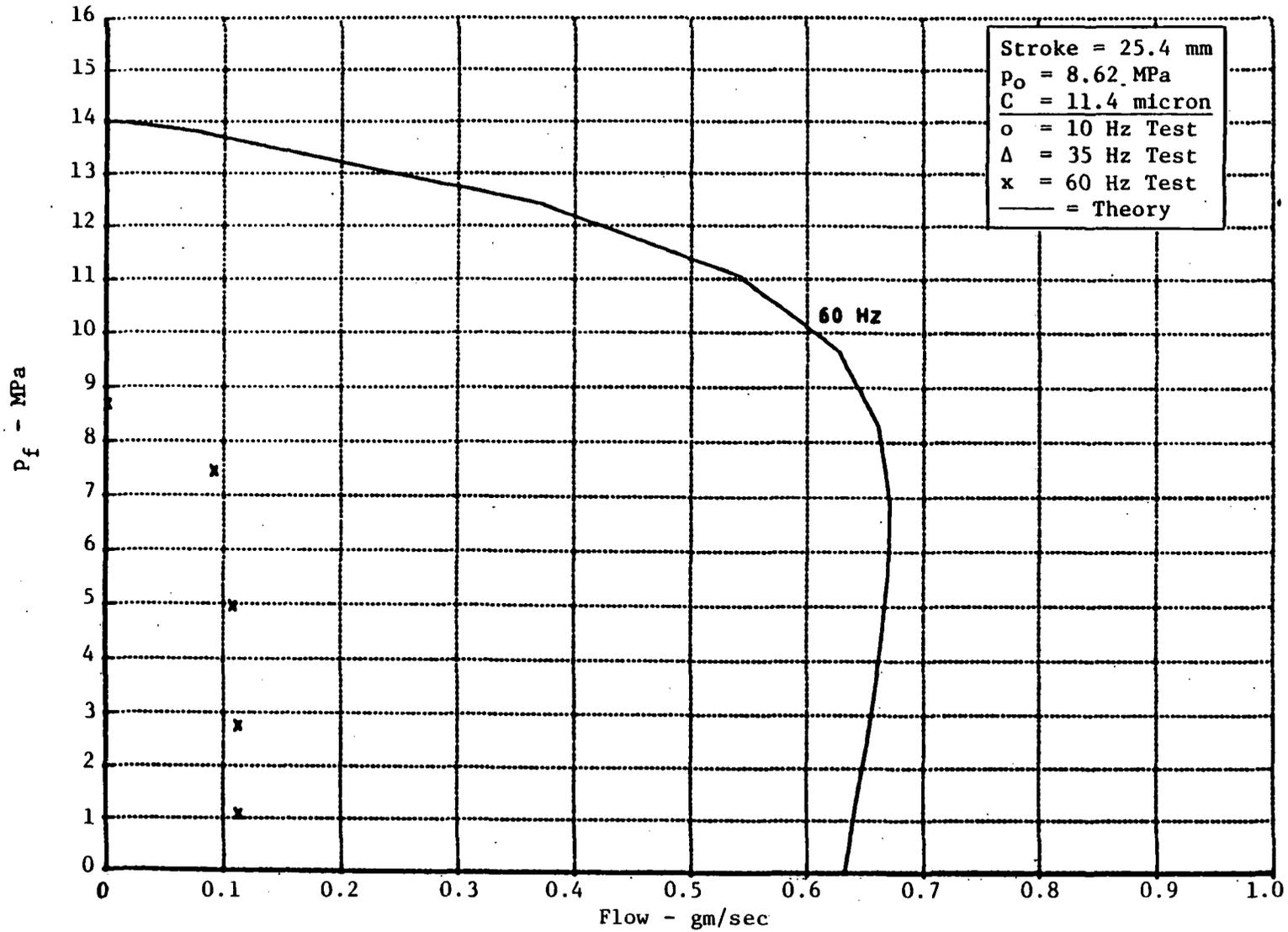


Fig. D-4 Performance of Babbitt Pumping Ring (I)

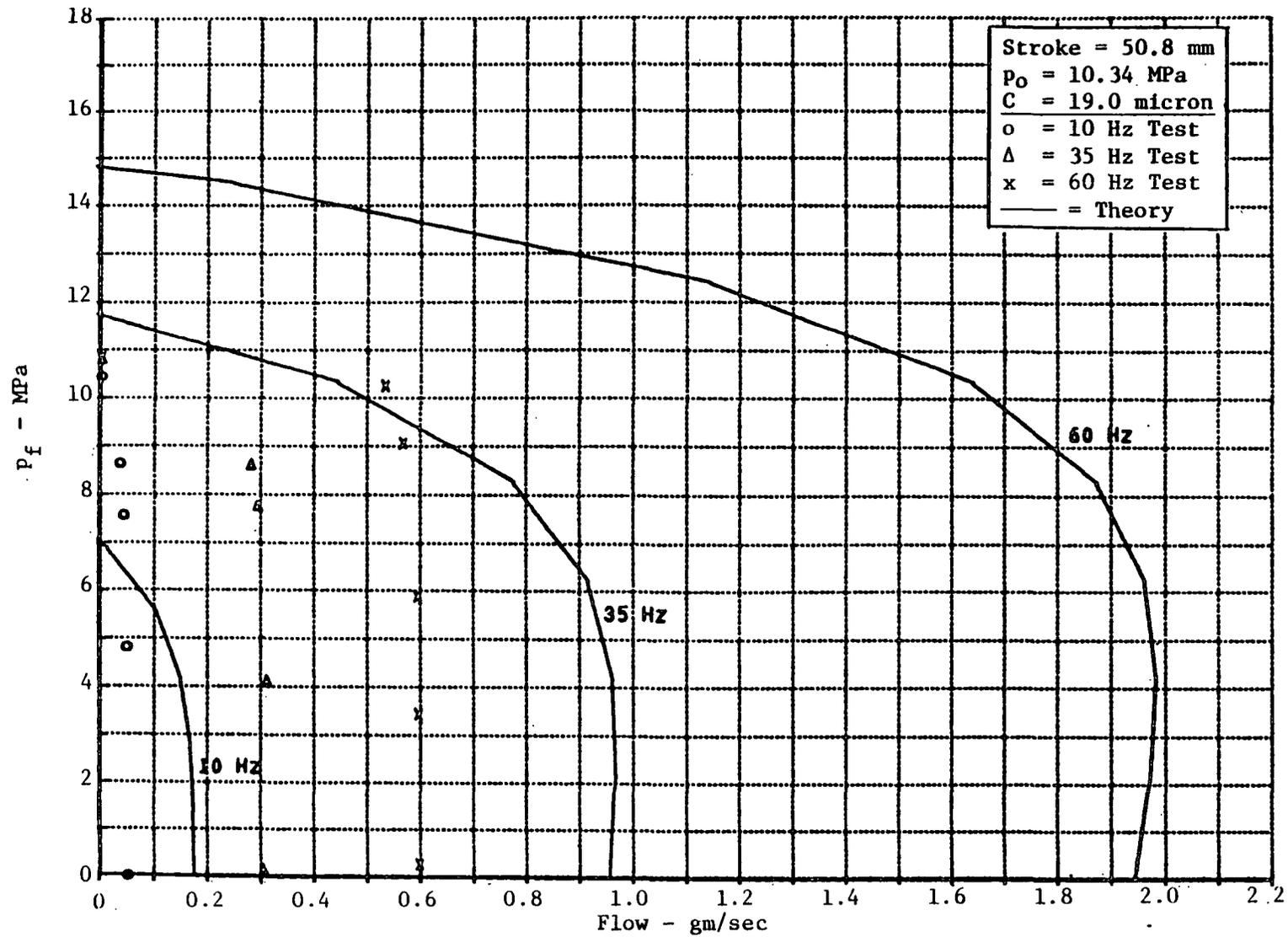


Fig. D-5 Performance of Babbitt Pumping Ring (II)

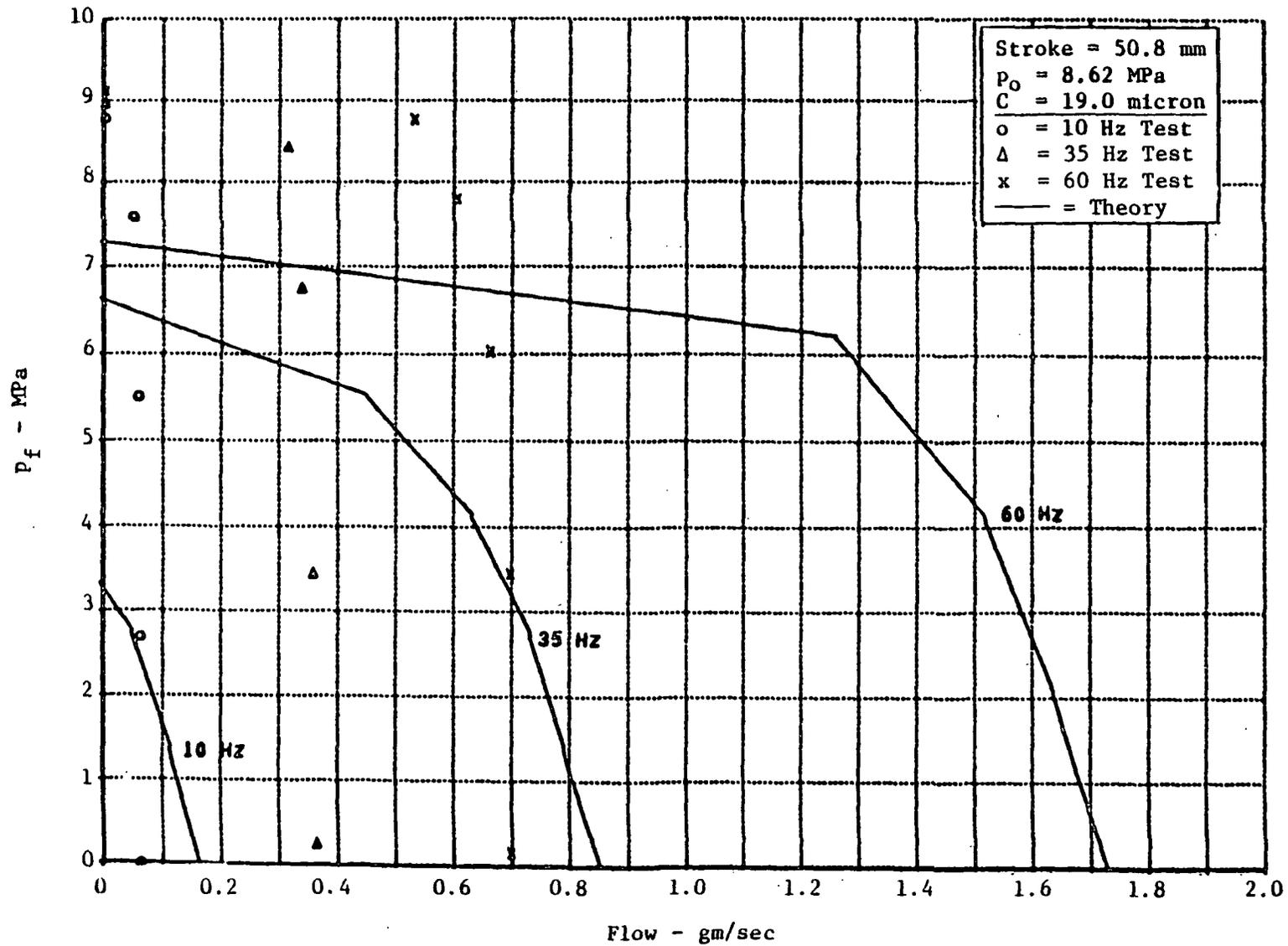


Fig. D-6 Performance of Babbitt Pumping Ring (II)

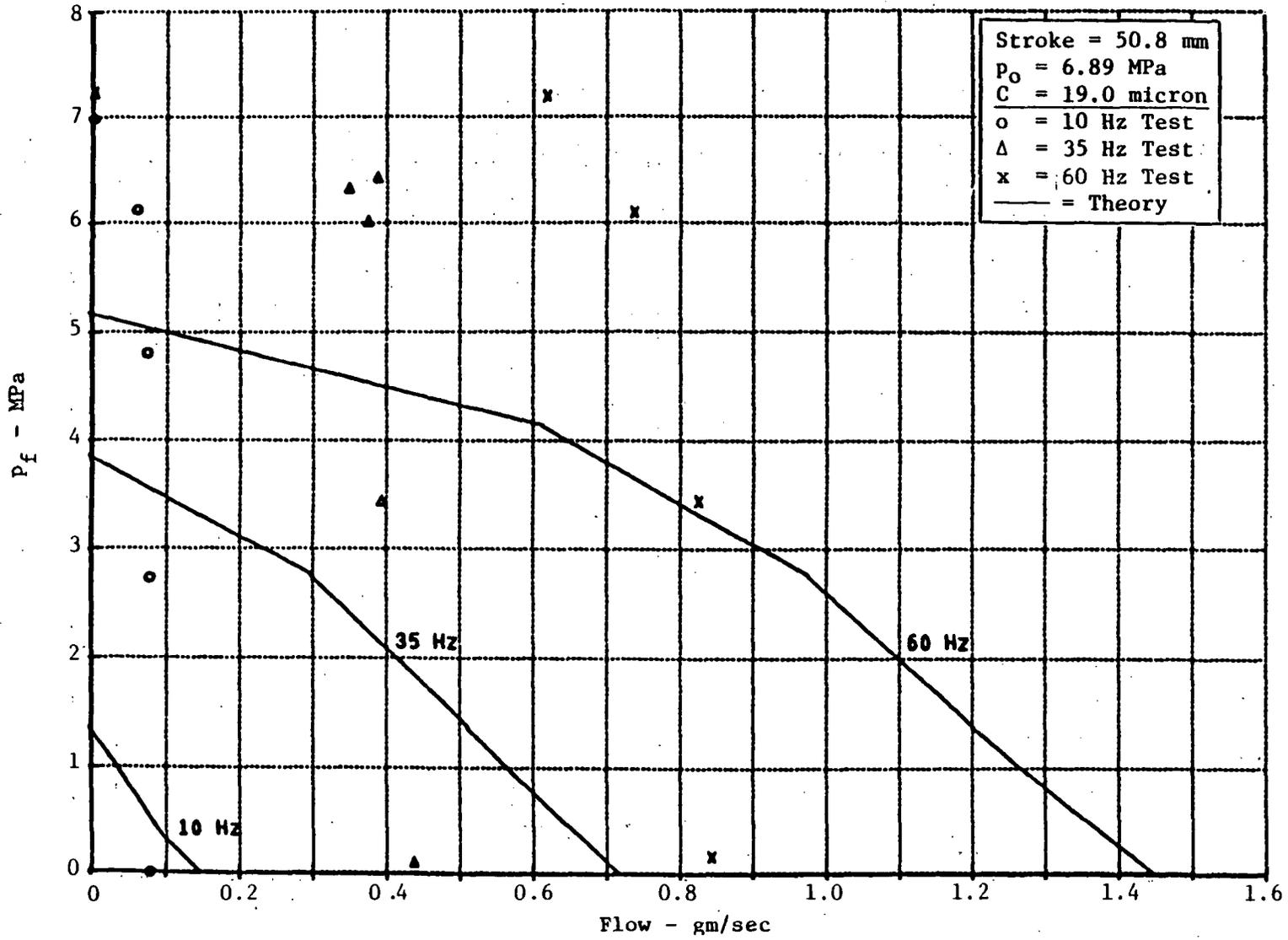


Fig. D-7 Performance of Babbitt Pumping Ring (II)

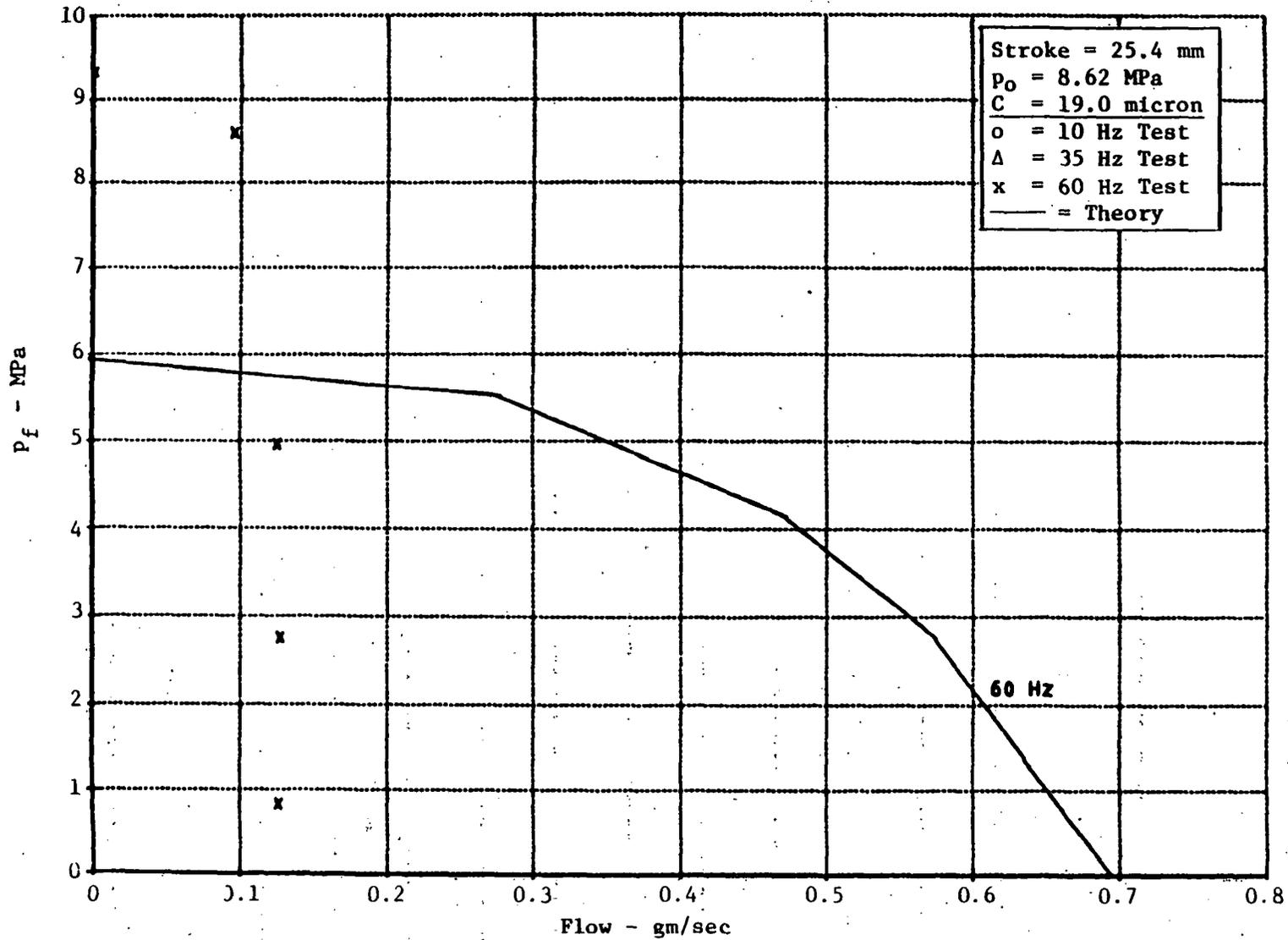


Fig. D-8 Performance of Babbitt Pumping Ring (II)

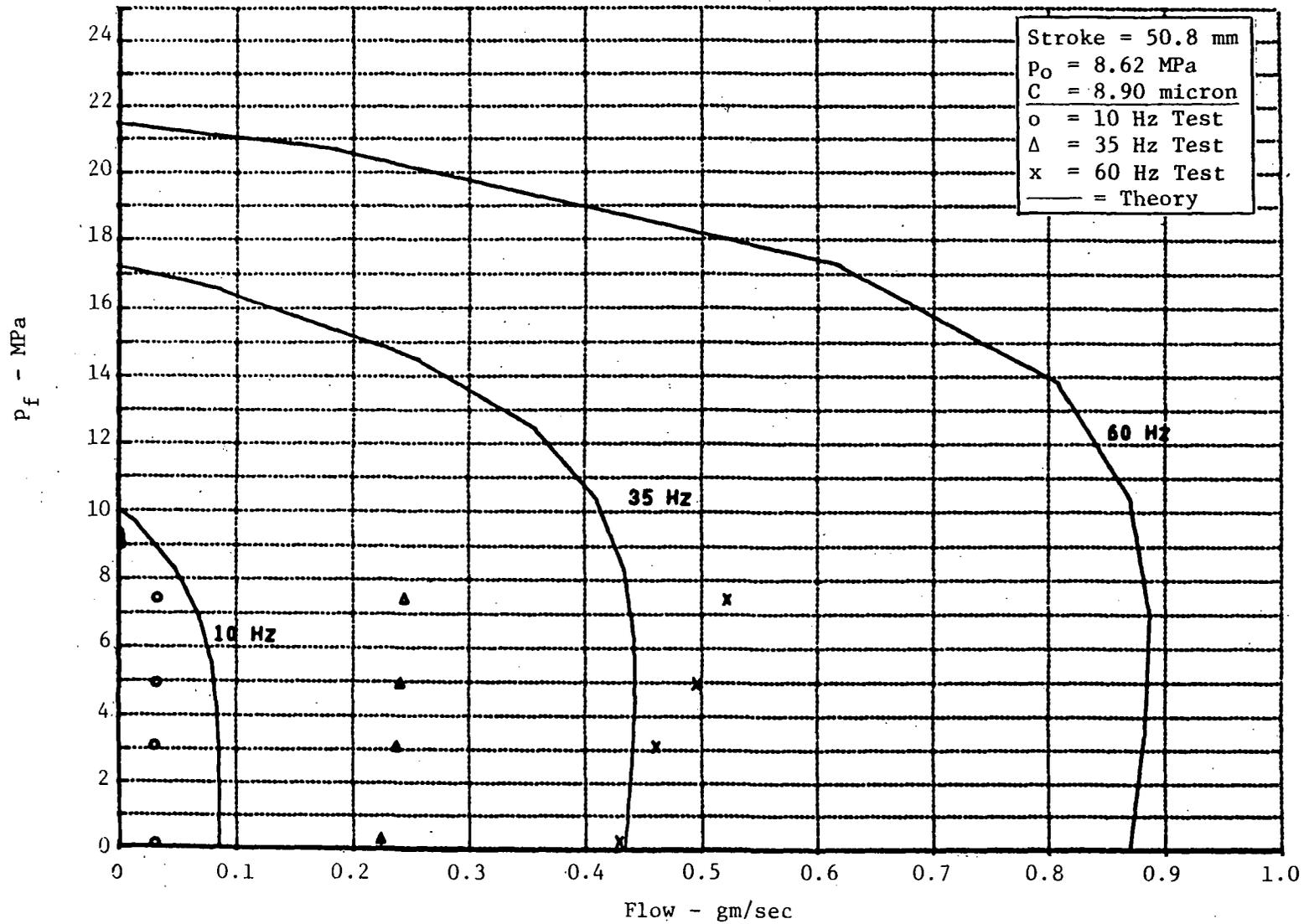


Fig. D-9 Performance of Small Babbitt Pumping Ring (VII)

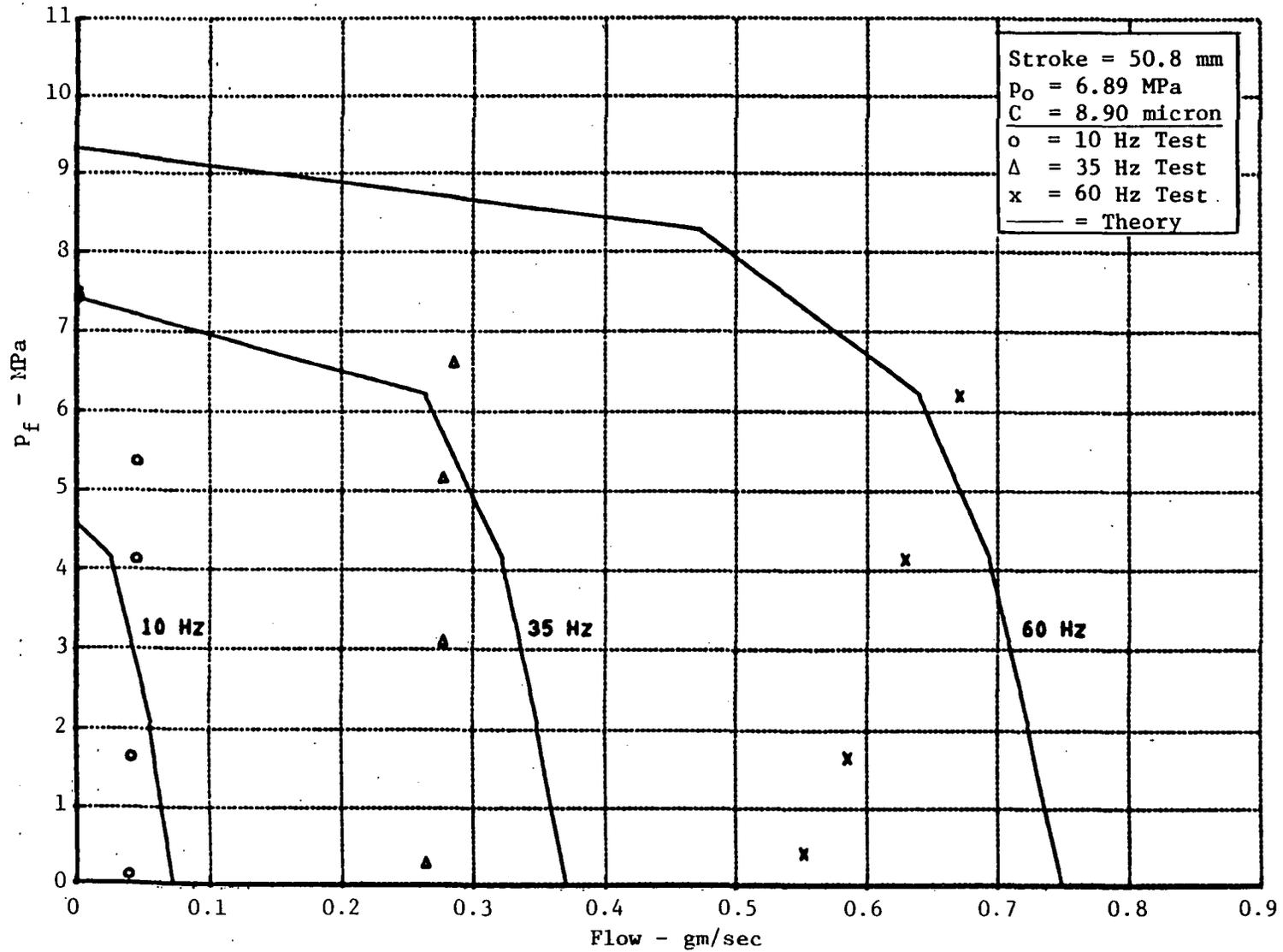


Fig. D-10 Performance of Small Babbitt Pumping Ring (VII)

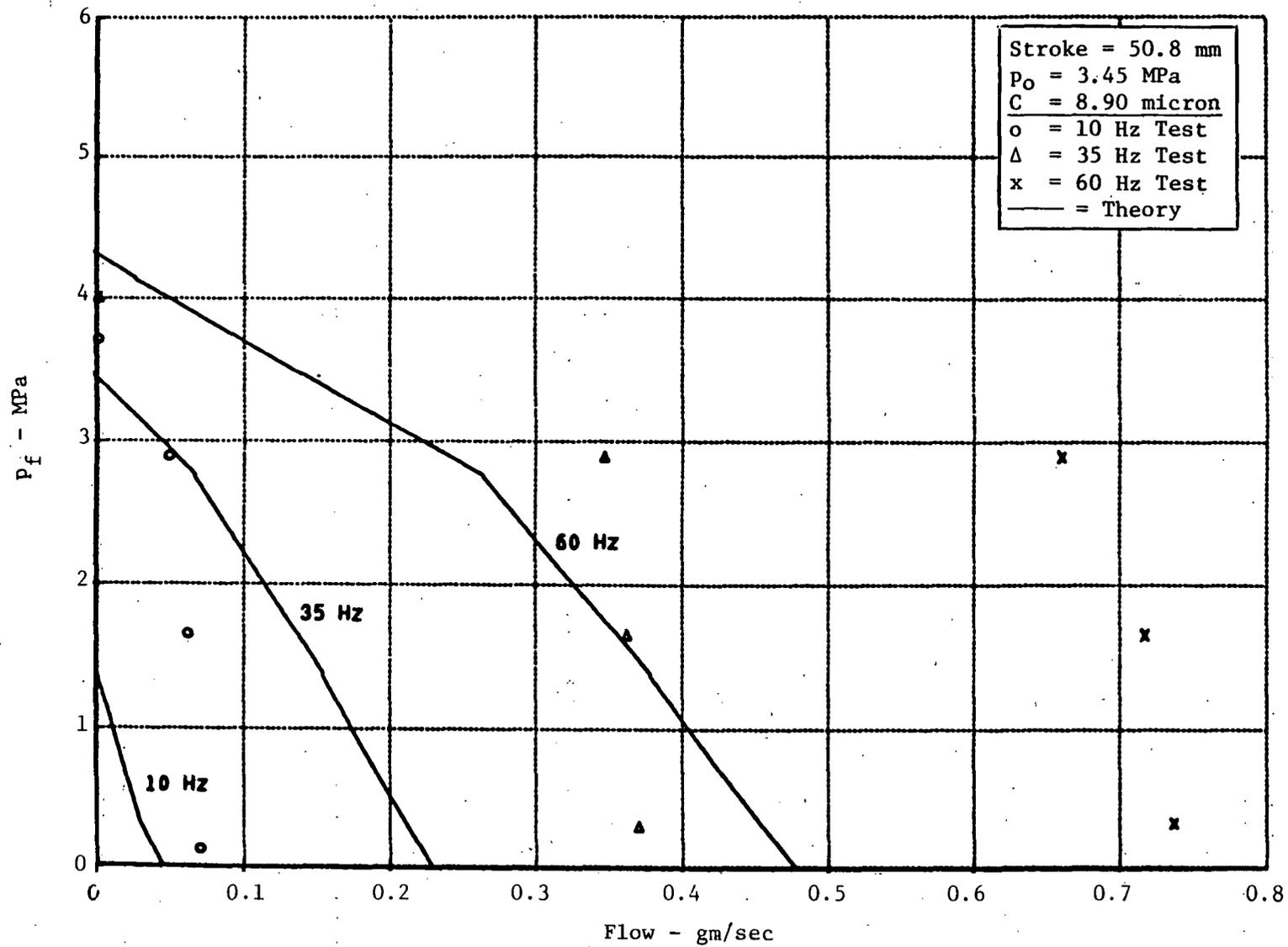


Fig. D-11 Performance of Small Babbitt Pumping Ring (VII)

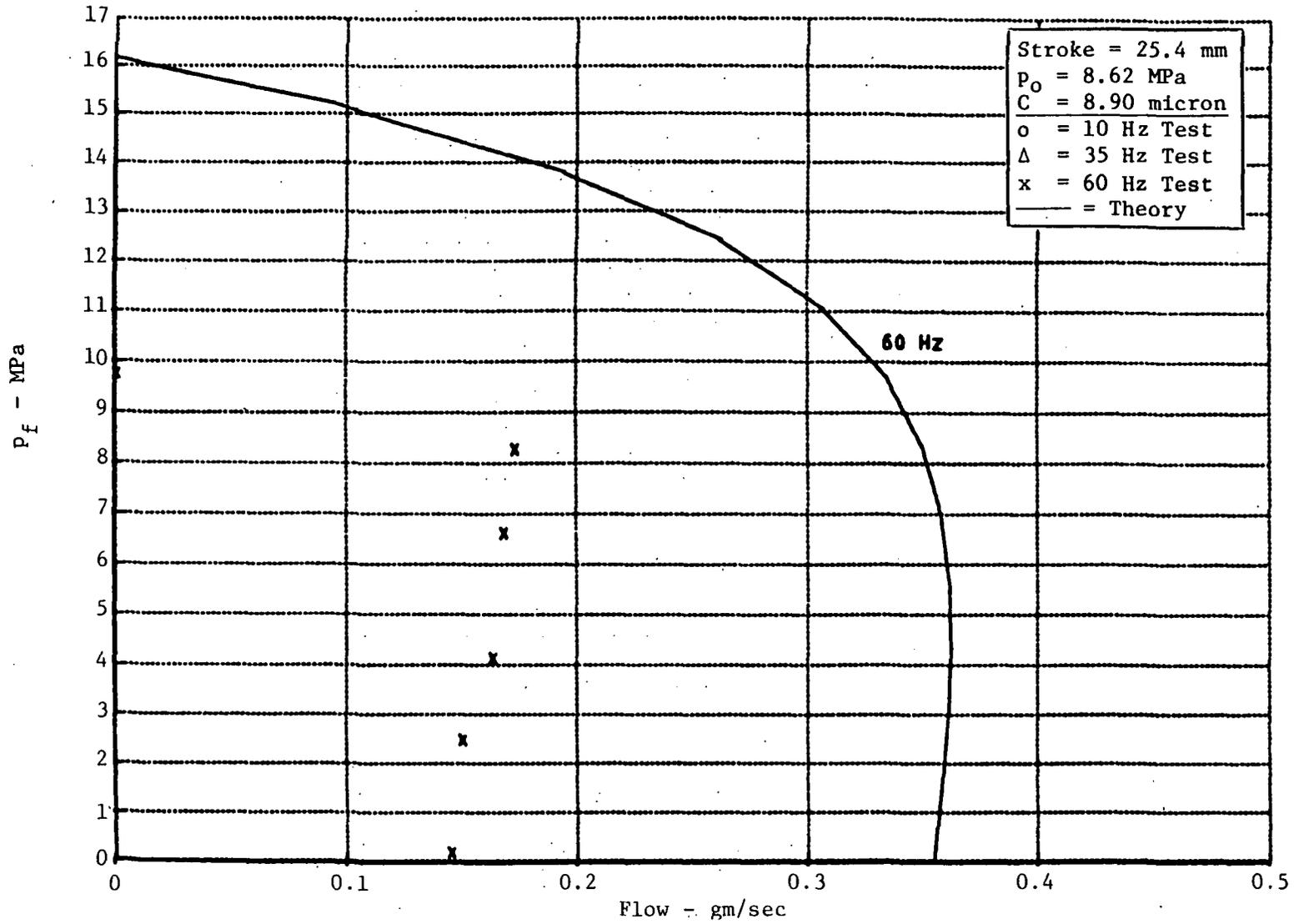


Fig. D-12 Performance of Small Babbitt Pumping Ring (VII)

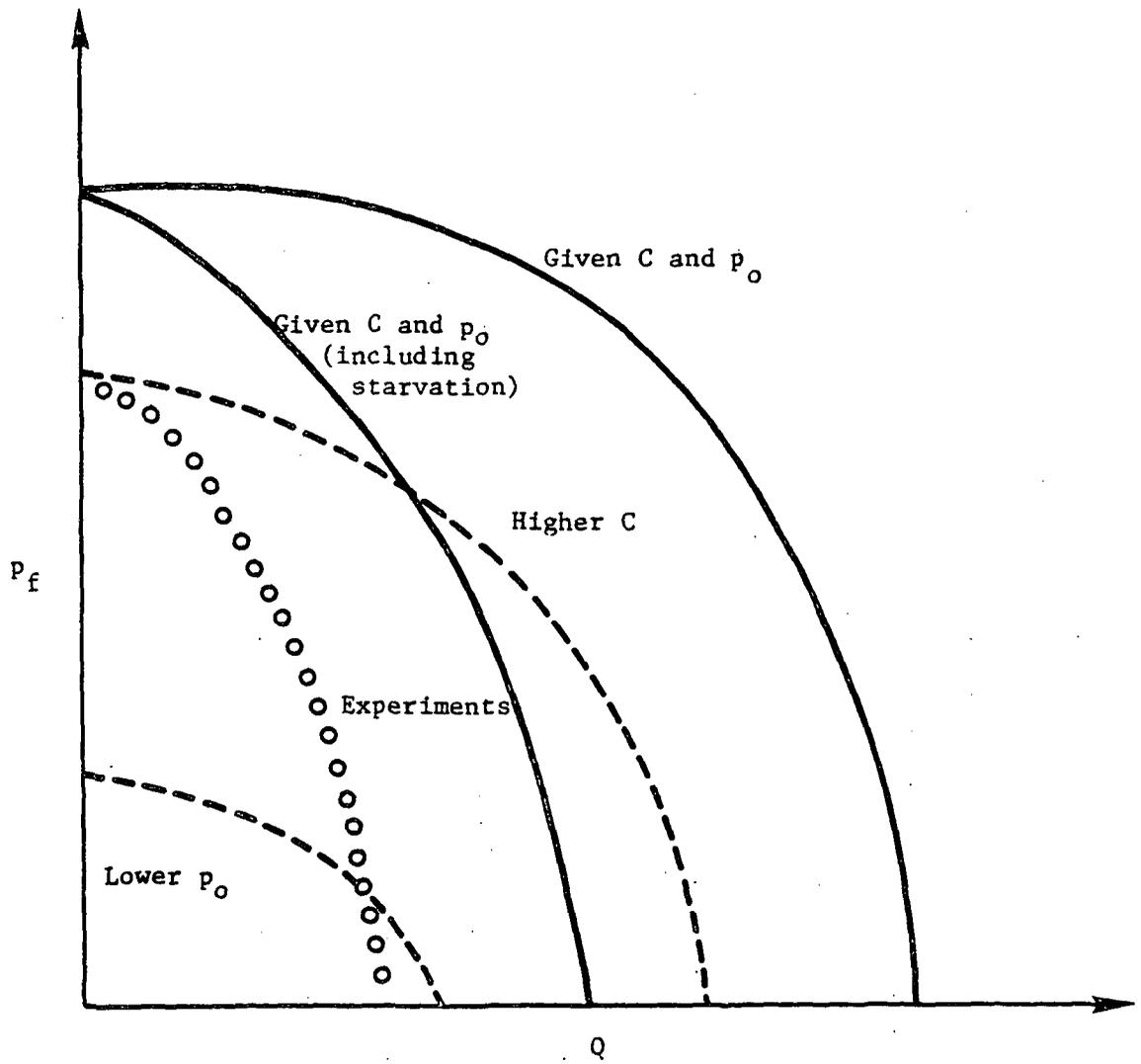


Fig. D-13 Qualitative Comparisons of Theoretical and Experimental Results

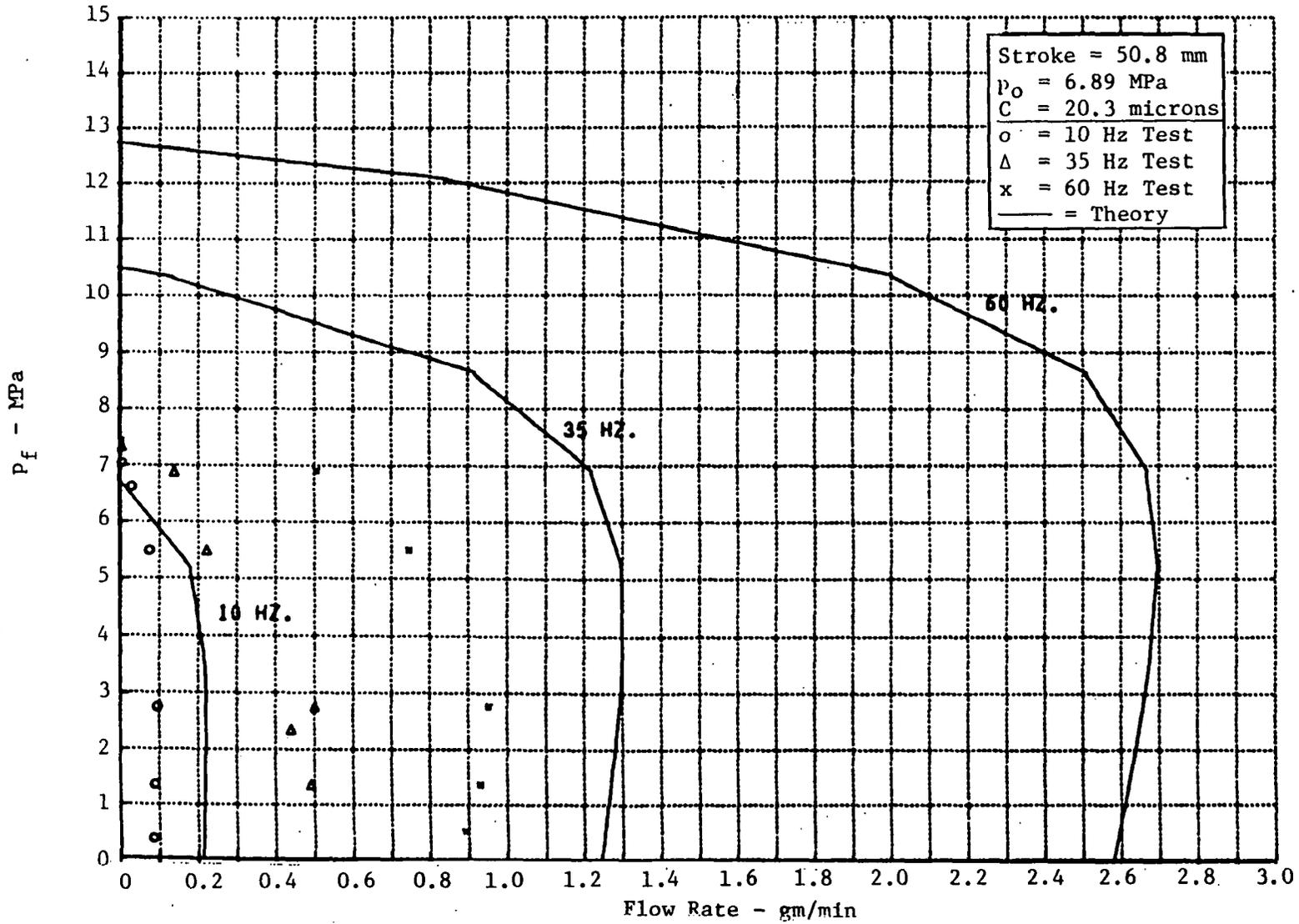


Fig. D-14 Performance of Carbon Pumping Ring

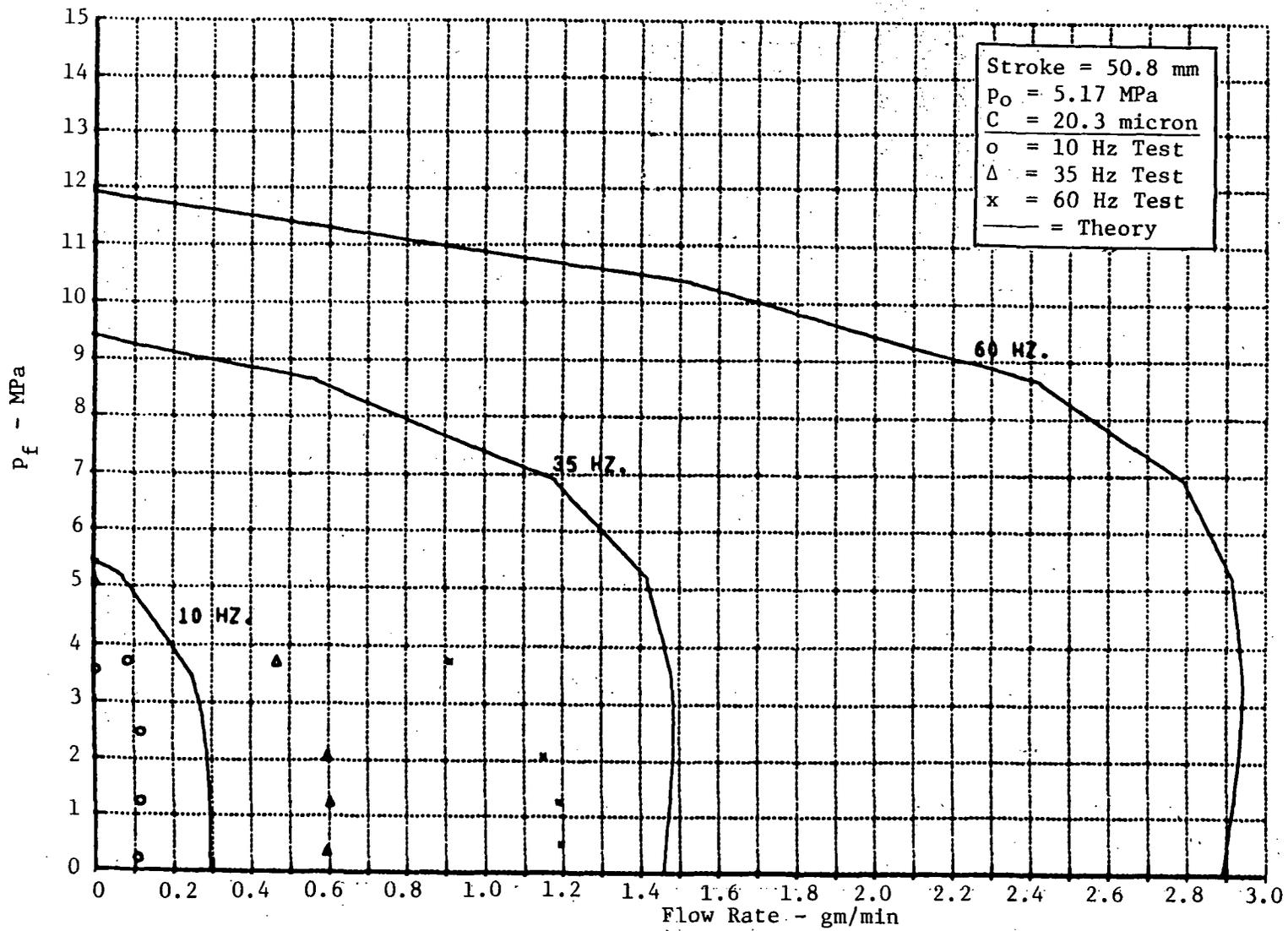


Fig. D-15 Performance of Carbon Pumping Ring

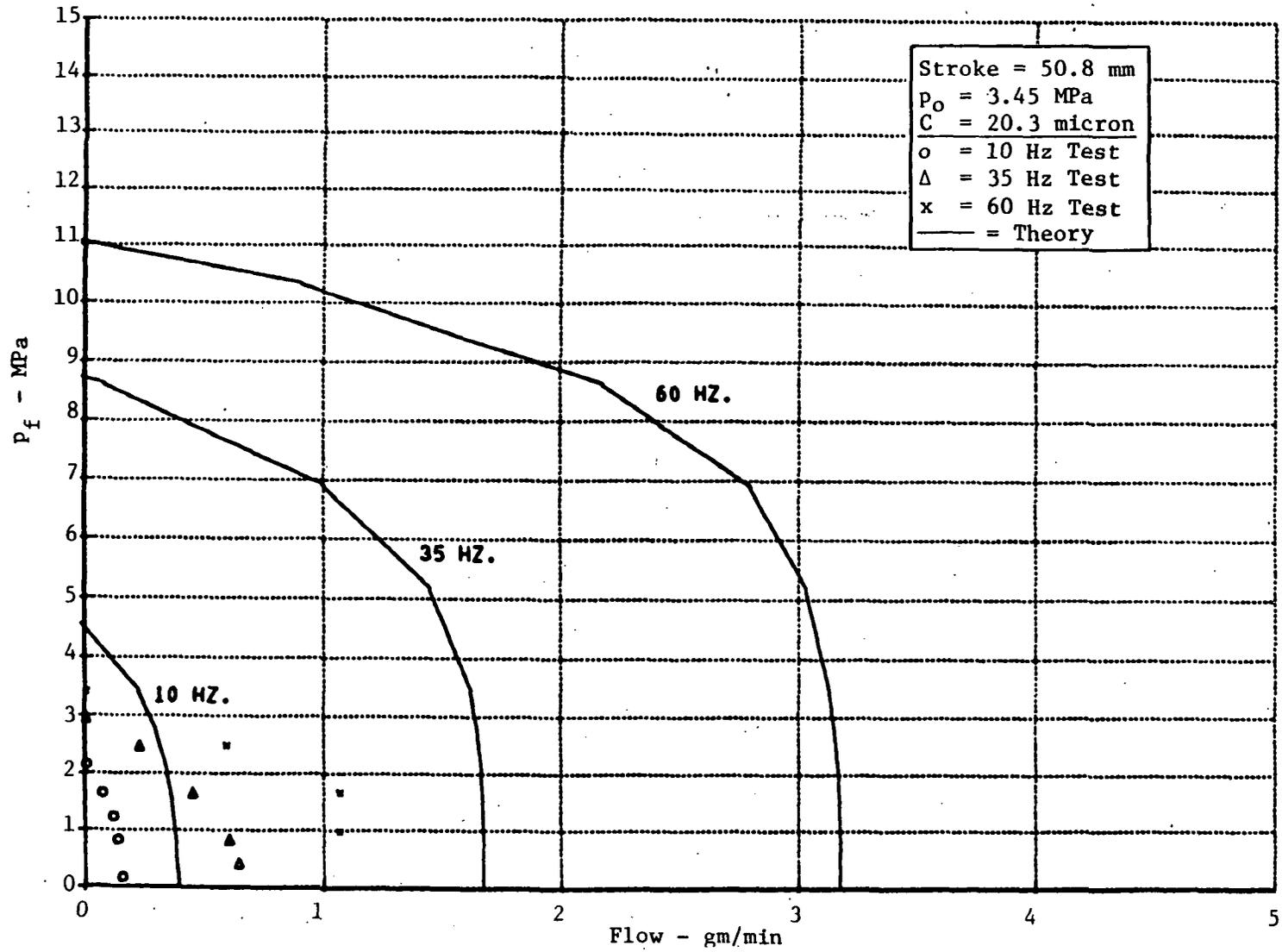


Fig. D-16 Performance of Carbon Graphite Pumping Ring

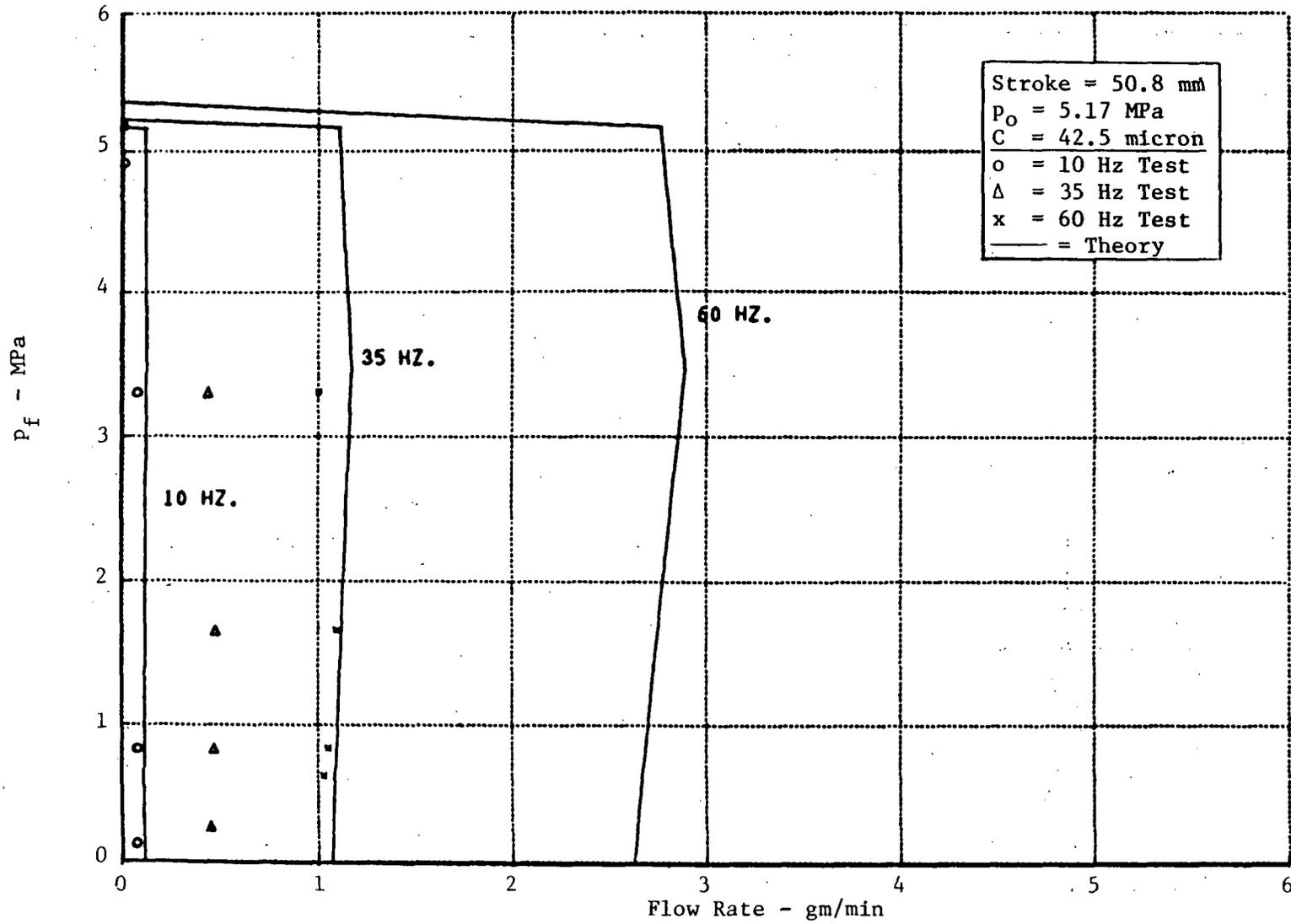


Fig. D-17 Performance of Rulon Pumping Ring

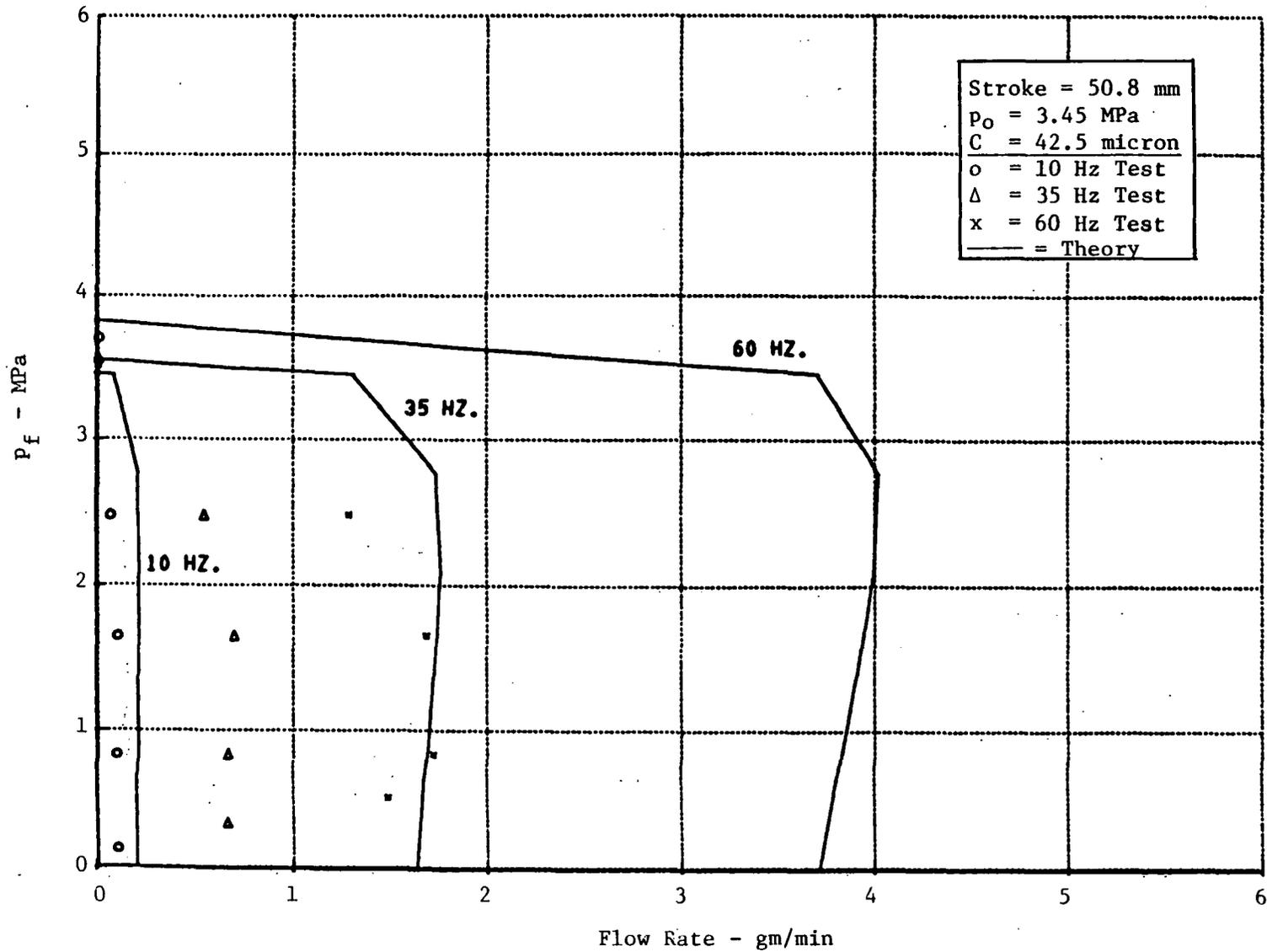


Fig. D-18 Performance of Rulon Pumping Ring

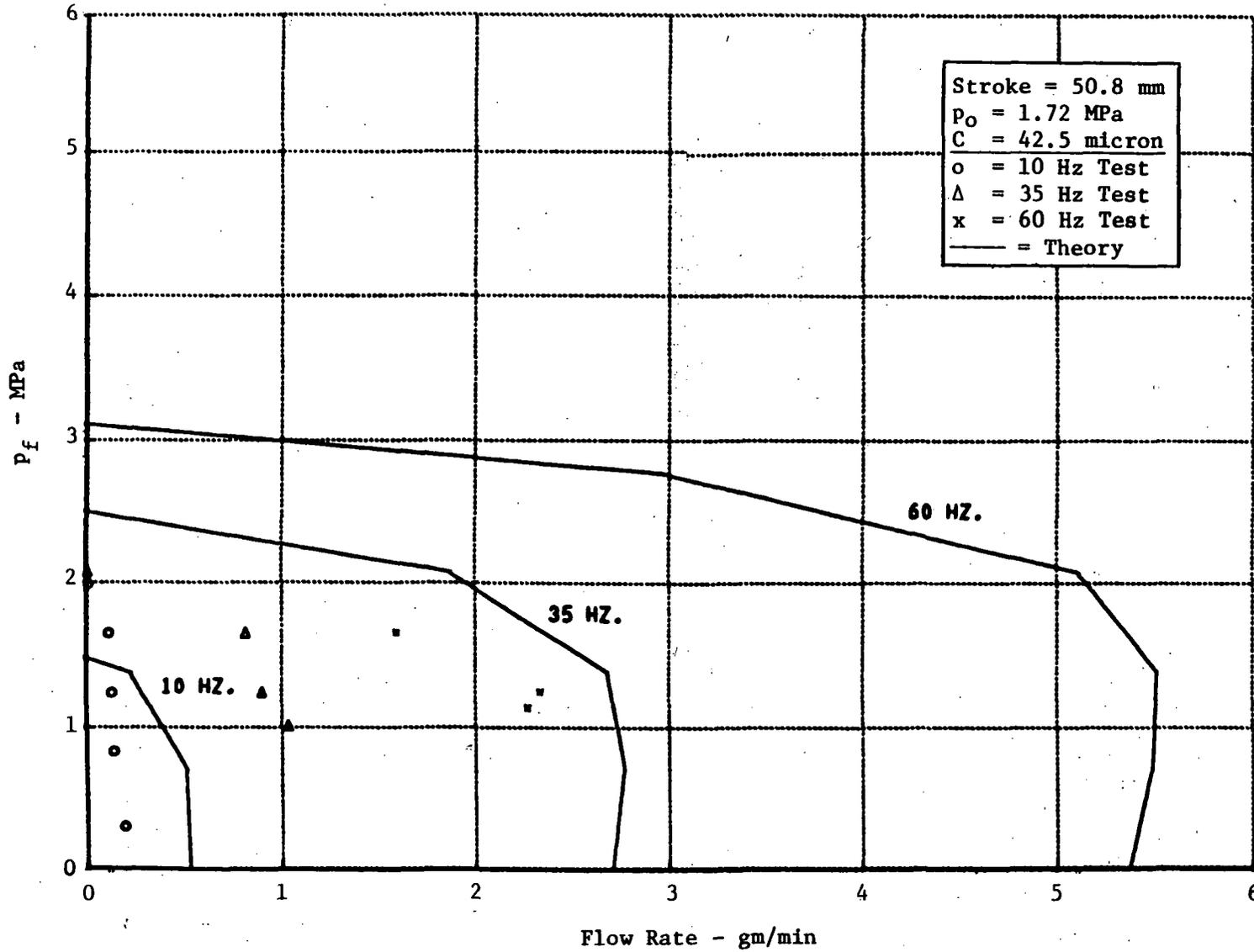


Fig. D-19 Performance of Rulon Pumping Ring

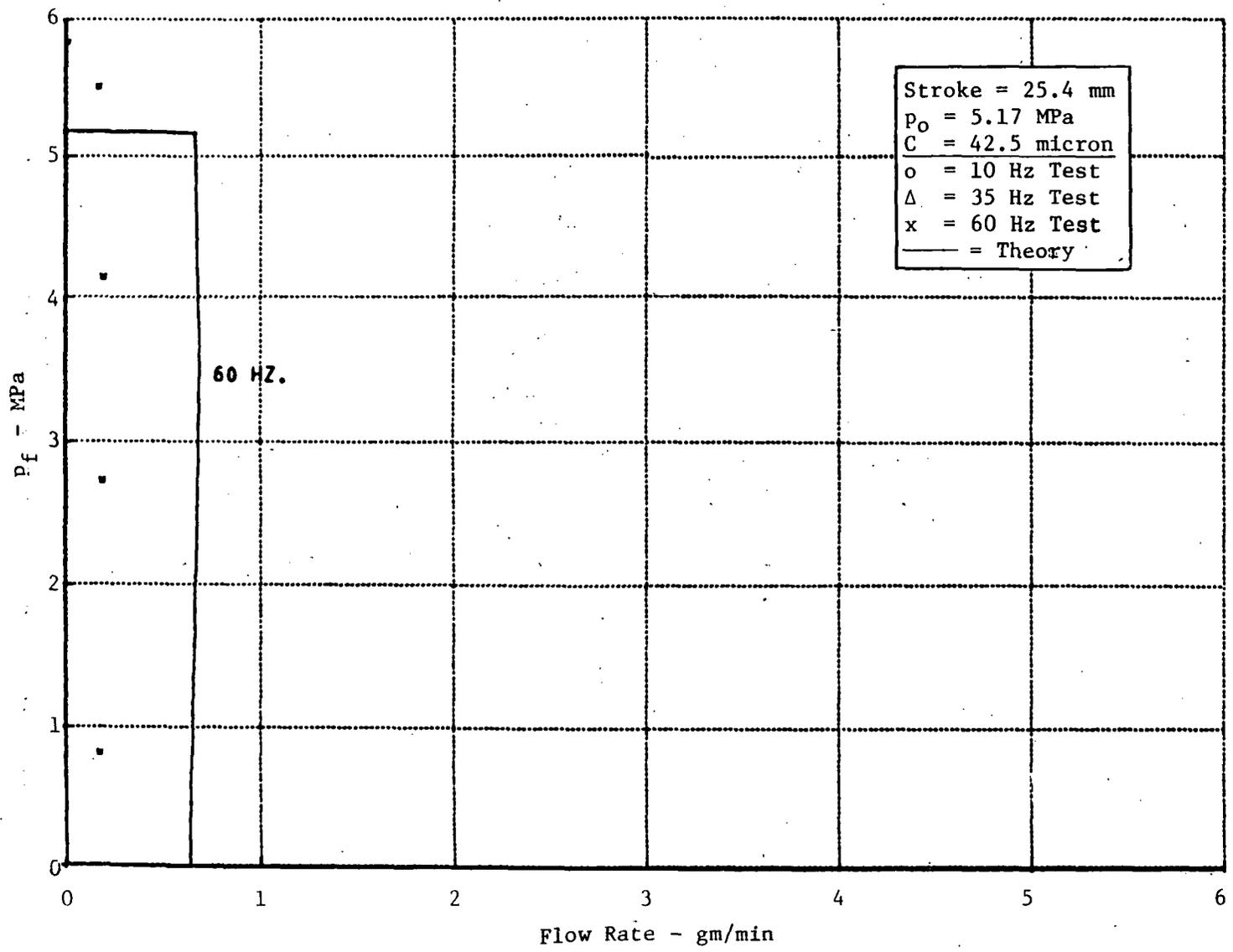


Fig. D-20 Performance of Rulon Pumping Ring

TABLE D-2

QUALITATIVE SUMMARY OF COMPARISON
BETWEEN ANALYTICAL RESULTS AND TESTS

RING		$(Q_o \text{ THEO}/Q_o \text{ EXP})$		$(P_{fm} \text{ THEO}/P_{fm} \text{ EXP})$	
		High p_o	Low p_o	High p_o	Low p_o
Babbitt	Small C	2	1	2	1
	Large C	3	1.5	1.5 - 0.5	0.7 - 0.2
	Small R	2	0.5	2	1 - 0.3
Carbon Graphite		2.5	3 - 2	1 - 2	1.5 - 4
Rulon		2.5	2.5	1	1.5 - 0.5

1. Report No. NASA CR-175083		2. Government Accession No.		3. Recipient's Catalog No.	
4. Title and Subtitle Performance of Oil Pumping Rings - An Analytical and Experimental Study				5. Report Date 1/17/86	
				6. Performing Organization Code	
7. Author(s) M.W. Eusepi, J.A. Walowit, O. Pinkus, and P. Holmes				8. Performing Organization Report No. MT1 86TR17.	
				10. Work Unit No.	
9. Performing Organization Name and Address Mechanical Technology Incorporated 968 Albany-Shaker Road Latham, New York 12110				11. Contract or Grant No. DEN 3-256	
				13. Type of Report and Period Covered Contractor Report	
12. Sponsoring Agency Name and Address U.S. Department of Energy Office of Vehicle and Engine R&D Washington, D.C. 20545				14. Sponsoring Agency Code Report No. DOE/NASA/0256-1	
15. Supplementary Notes Final Report. Prepared under Interagency Agreement DE-A101-85CE50112. Project Manager, William Tomazic, Power Technology Division, NASA Lewis Research Center, Cleveland, Ohio 44135.					
16. Abstract An analysis, computer program and experimental data have been generated to provide design data for pumping rings. Although some disagreement has been found to occur in flow prediction, a procedure has been used to design pumping rings that performed well experimentally. A preliminary analysis has also been developed for the prediction of film thickness and pressure distribution in a pumping Leningrader seal.					
17. Key Words (Suggested by Author(s)) Hydrodynamic oil pumping ring; Stirling engine rod seal; Reciprocating rod seal analysis; Reciprocating rod seal design			18. Distribution Statement Unclassified - unlimited STAR Category 34 DOE Category UC-96		
19. Security Classif. (of this report) Unclassified		20. Security Classif. (of this page) Unclassified		21. No. of pages 233	22. Price* A11