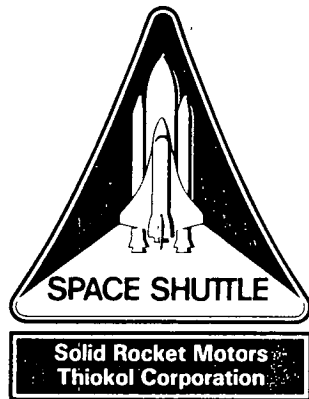


CR-184001



TWR-17270 VOL. VI

QM-6 FINAL PERFORMANCE EVALUATION REPORT
IGNITER

JULY 1990

Prepared for:

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION
GEORGE C. MARSHALL SPACE FLIGHT CENTER
MARSHALL SPACE FLIGHT CENTER, ALABAMA 35812

Contract No. NAS8-30490

DR. No. 5-3

WBS.No. 4B102-10-06

ECS. 1002

Thiokol CORPORATION
SPACE OPERATIONS

P.O. Box 707, Brigham City, UT 84302-0707 (801) 863-3511

(NASA-CR-184001) QM-6 FINAL PERFORMANCE
EVALUATION REPORT IGNITER, VOLUME 6
(Thiokol Corp.) 48 p

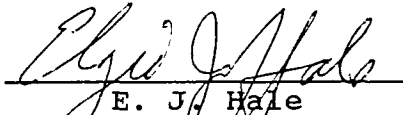
N90-27779

CSCL 21H

Unclas
63/20 0297557

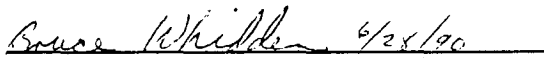
TWR-17270
QM-6 FINAL PERFORMANCE EVALUATION REPORT
IGNITER

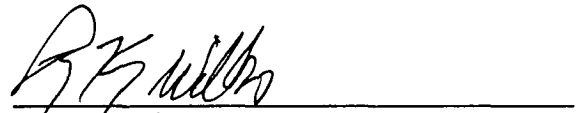
PREPARED BY:


E. J. Hale

Ignition, Instrumentation and Electrical Design

APPROVED BY:


Bruce Whidden, Supervisor
Ignition, Instr. and Elect.


R. K. Wilks, Manager
Non Metallic Component Design


T. M. Gregory
Project Engineering


Chris Rice
Project Management


SR & QA

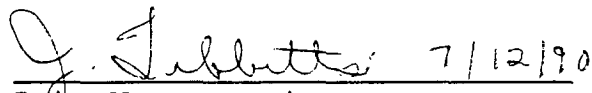

Data Management

TABLE OF CONTENTS

<u>SECTION</u>	<u>DESCRIPTION</u>	<u>PAGE</u>
1.0	Introduction	1
2.0	Objectives	1
3.0	Applicable Documents	2
4.0	Summary/Conclusions	2
4.1	Igniter Performance	2
4.2	S&A Device	5
5.0	Results/Discussion	5
5.1	Sealing System	5
5.1.1	S&A Device to Adapter	5
5.1.2	Igniter Chamber to Adapter	5
5.1.3	Igniter Joint Adapter-to-Forward Dome	13
5.2	Insulation	13
5.3	Nozzle Insert	13
6.0	Metal Components	22
7.0	Ignition System Component Team Recommendations	22
8.0	Igniter Pedigree and Measured Performance	22
Appendix A:	Evaluation Checkoff Worksheets	23

LIST OF FIGURES

<u>Figure</u>	<u>Description</u>	<u>Page</u>
Figure 1	Igniter Cross Section	3
Figure 2	QM-6 Igniter Performance and Specification Limits . .	4
Figure 3	S&A Sealing Surface	9
Figure 4	S&A Adapter Sealing Surface	10
Figure 5	Igniter Chamber	11
Figure 6	Igniter Gask-O-Seal	12
Figure 7	Igniter prior to Removal	14
Figure 8	Igniter Adapter	15
Figure 9	Polar Boss	16
Figure 10	Igniter Adapter Sealing Surface	17
Figure 11	Igniter Insulation Repair Patch	18
Figure 12	Test Plug Position for Insulation Measurements . .	19
Figure 13	Nozzle Insert	21

LIST OF TABLES

<u>Table</u>	<u>Description</u>	<u>Page</u>
Table I	Qualification Objectives and Results	6
Table II	Igniter Insulation History	20

1.0 INTRODUCTION

The Space Shuttle Redesigned Solid Rocket Motor (RSRM) static test of Qualification Motor (QM-6) was conducted 20 April 1988 at Morton Thiokol, Inc., Space Operations. The QM-6 was a full scale, full duration test.

This report does not include all test results, but rather, addresses the performance of the igniter, safing and arming device, and their associated seals.

2.0 OBJECTIVES

The objective of this report is to document the post test condition of the Igniters and associated components, and to show QM-6 static test compliance with Specification CPW1-3600, Addendum D, dated 3 Aug 1987.

Qualification Objectives (CEI Paragraph)

- G. Certify that all RSRM seals experience no erosion or blowby through the static test (3.2.1.2).
- V. Verify that the ignition system seals, if pressurized, accommodate static test motor structural deflections (3.2.1.2.4.a).
- W. Certify that the ignition system seals, if pressurized, operate at test temperature, 55° minimum (3.2.1.2.4.b).
- X. Certify that the ignition system seal verification does not degrade the performance of integrity of the sealing system (3.2.1.2.4.c).
- AR. Certify that the ignition system precludes hot gas leakage during and subsequent to motor ignition (3.2.1.5.a).
- AS. Certify that the igniter and S&A are separable from each other (3.2.1.5.b).
- AT. Certify the enable function of the S&A device (3.2.1.5.1.a).
- AU. Certify the S&A change of position from safe to arm. (3.2.1.5.1.d).
- AV. Certify that the S&A device will provide simplex remote position indication (3.2.1.5.1.e).
- AW. Certify the igniter design (3.2.1.5.2)

- BF. Certify by demonstration the field joint and igniter to case joint leak test compatibility. (3.2.1.8.1.1.b).
- BG. Certify that the field joint and the igniter to case joint insulation provide seal protection. (3.2.1.8.1.1.d).
- BH. Certify the field joint and igniter to case joint insulation performance during and ambient temperature full duration motor burn. (3.2.1.8.1.1.e).
- BI. Certify by inspection that the field joint, case to nozzle and igniter to case joint insulation will withstand slag accumulation during motor operation. (3.2.1.8.1.1.g).
- BK. Certify that the igniter insulation provides adequate thermal protection for the main igniter chamber and adapter metal parts (3.2.1.8.3).

3 .0 APPLICABLE DOCUMENTS

1U75166 Igniter Rocket Motor-Test Configuration,
Modified

TWR-16473 Vol VI Qualification and Production Motor
Postfire engineering Plan (Igniter Component)

CTP-0028 Space Shuttle Qualification Motor # 6 (QM-6)
Static Fire Test Plan

4.0 SUMMARY/CONCLUSIONS

4.1 Igniter Performance

The overall performance of the igniter components was excellent. Figure 1 illustrates the Igniter and S & A seals and components. The igniter performance was within specified limits as shown in Figure 2. No damage or heat affected areas were noted except for minor seal damage which probably occurred during disassembly.

The sealing elements of the igniter functioned as expected with no evidence of erosion or blowby.

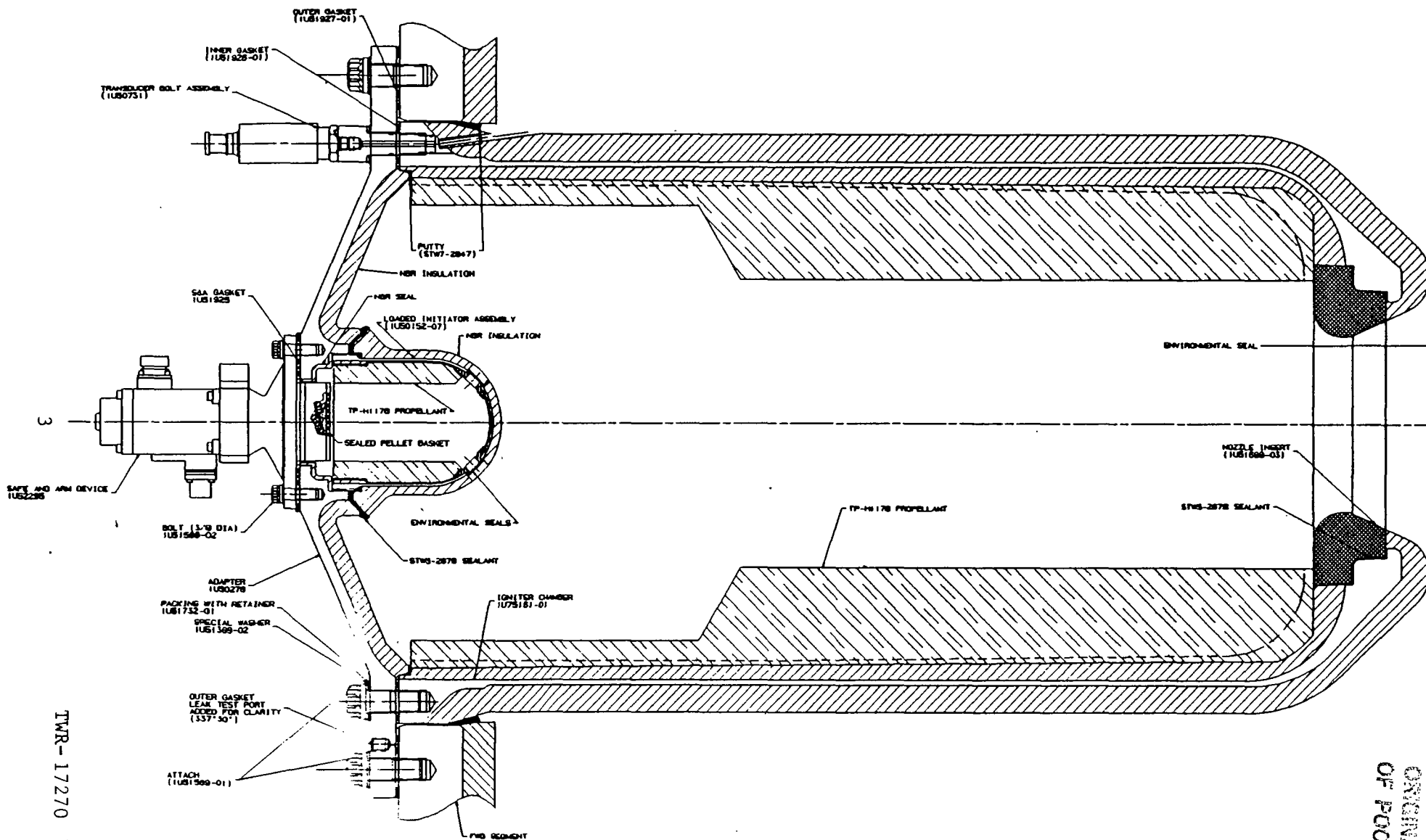


FIGURE 1. IGNITER CROSS SECTION

ORIGINAL PAGE IS
OF POOR QUALITY

QM-6 IGNITER RECONSTRUCTION AT 80 DEG F

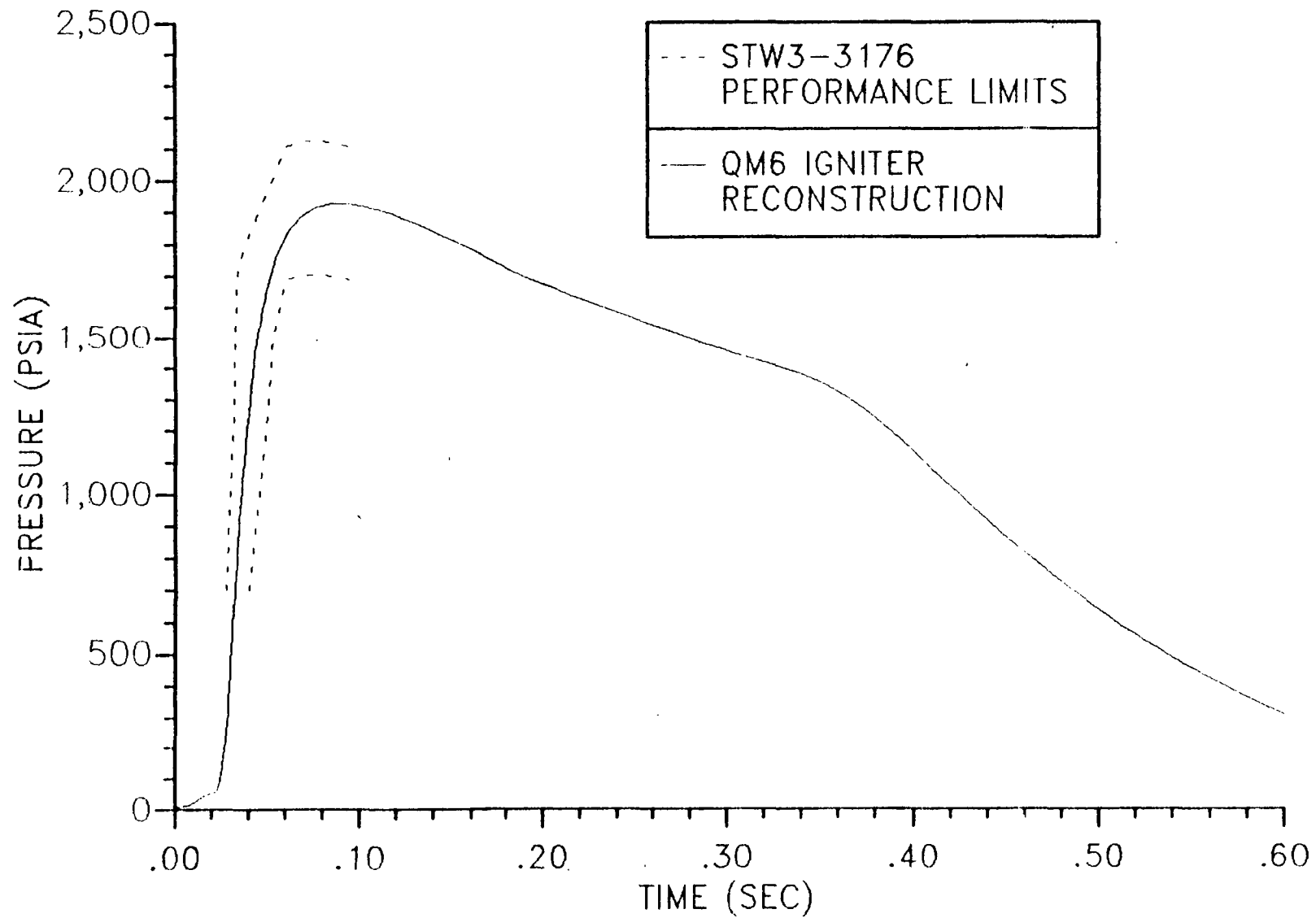


FIGURE 2. IGNITER PERFORMANCE & SPECIFICATION LIMITS

The condition of the igniter to case joint insulation was excellent. The igniter boss insulation exhibited normal erosion on the inboard surface. No edge separations were identified. The igniter chamber interior and exterior insulation was in normal condition. One blowhole through the putty was present, however, no adverse effects on the performance of the joint resulted from the blowhole.

See Table I for a tabulated summary of a one-to-one correlation of the test objectives listed in Section 2 with the observed results.

4.2 S&A Device.

The S&A enabled the motor ignition sequence, as evidenced by the successful firing. The S&A was separated from the QM-6 igniter during disassembly at T-24, demonstrating separability.

During the motor firing, the S&A provided simplex position indication to the control center, allowing continuation of the motor countdown.

5.0 RESULTS/DISCUSSION

5.1 SEALING SYSTEM

5.1.1 Safe and Arm Device to Adapter

The Safe and Arm-to-igniter joint was disassembled on 5 May 1988. The S&A gasket and both S&A and igniter adapter sealing surfaces were thoroughly inspected, revealing no damage, erosion, or heat effect (see Figures 3 and 4). There was no soot to, or past the primary seal. The sealing surfaces of the joint were in excellent condition (see page A-1).

At 225 degrees, a depression of approximately 0.25 in. long, and 0.015 in. deep was found in the crown of the primary seal of the S&A gasket (see pages A-3 through A-5). Thiokol laboratory analysis showed that a combination of mold defect and gas entrapment were the cause of the depression.

5.1.2 Igniter Chamber to Adapter

The igniter adapter-to-chamber joint was disassembled on 3 June 1988. Inspection of the inner gasket revealed no damage, heat effect, or soot to or past the primary seal (see Figures 5 and 6). Light soot was found on the outside edge of the gasket, covering the entire circumference. The sealing surfaces were free of soot and in excellent condition (see pages A-6 through A-11).

TABLE I
QUALIFICATION OBJECTIVES AND RESULTS

<u>Qualification Objectives (CEI paragraph number)</u>	<u>Pass/Fail Criteria</u>	<u>Results (Reference Section)</u>
G Certify that all RSRM seals, including adjustable vent port plug seals, experience no erosion or blowby throughout the static test. (3.2.1.2)	There must be no erosion or blowby for the specified seals (acceptable erosion is allowed on factory joint insulation acting as the primary seal).	No evidence of erosion or blowby past the RSRM seals. (5.1)
V Certify that the ignition system seals, if pressurized, accommodate static test motor structural deflections. (3.2.1.2.4.a)	There must be no erosion or gas leakage past the ignition systems seals due to structural deflections.	No evidence of seal failure. (5.1)
W Certify that the ignition system seals, if pressurized, operate at test temperature, 55° minimum. (3.2.1.2.4.b)	There must be no erosion or gas leakage past the ignition systems seals.	Disassembly showed seals operated correctly and as designed. (5.1)
X Certify that the ignition system seal verification does not degrade the performance or integrity of the sealing system. (3.2.1.2.4.c)	There must be no damage to the ignition system seals due to leak check procedures.	Disassembly verified that seal performance is not degraded. (5.1)
AR Certify that the ignition system precludes hot gas leakage during and subsequent to motor ignition. (3.2.1.5.a)	There must be not hot gas leakage.	Disassembly verified that hot gas leakage was precluded. (5.1)
AS Certify that the igniter and the S&A are separable from each other. (3.2.1.5.b)	Separation of the igniter and S&A will be demonstrated.	The igniter and S&A were separated during teardown, demonstrating that the S&A can be removed from the igniter during routine disassembly. (5.1)

AT	Certify the enable function of the S&A device. (3.2.1.5.1.a)	S&A device barrier-booster assembly must provide the energy flow required to achieve igniter initiation upon command.	The S&A device both enabled and inhibited ignition upon command. (4.2)
AU	Certify the S&A change of position from safe to arm and arm to safe. (3.2.1.5.1.d)	S&A will change position from safe to arm and arm to safe.	The S&A changed position from safe to arm and arm to safe upon command. (4.2)
AV	Certify that the S&A device will provide simplex remote position indication. (3.2.1.5.1.e)	S&A device must provide remote indication of position in both the safed and armed positions.	The S&A provided remote indication of position. (4.2)
AW	Certify the igniter design. (3.2.1.5.2)	There must be no debris formed that can damage any other component which is attributable to the igniter, and the igniter must be installed in only one predetermined rotational position into the igniter port from the outside of the forward segment.	Igniter operated as designed. (4.1)
BG	Certify that the field joint, case-to-nozzle joint and igniter-to-case joint insulation provide seal protection. (3.2.1.8.1.1.d)	There must be no erosion or hot gas jetting on the field joint, case-to-nozzle joint or igniter-to-case joint primary and secondary seals.	No evidence of erosion or hot gas jetting on the field joint or case-to-nozzle joint primary and secondary seals. (5.1.2) Igniter-to-case joint information will be included in TWR-17272, Vol II.
BH	Certify the field joint and igniter-to-case joint insulation performance during an ambient temperature full duration motor burn. (3.2.1.8.1.1.e)	There must be no thermal degradation to the field joint, case-to-nozzle joint and igniter-to-case joint.	No evidence of thermal degradation to the field joint or case-to-nozzle joint. (5.2)

BI Certify by inspection that the field joint, case-to-nozzle joint and igniter-to-case joint insulation will withstand slag during motor operation.
(3.2.1.8.1.1.g)

Slag accumulation must not cause insulation to fail to provide adequate protection of metal parts.

No evidence that slag adversely affected insulation. Insulation information is included in TWR-17272, Vol III.

BK Certify that the igniter insulation provides adequate thermal protection for the main igniter chamber and adapter metal parts.
(3.2.1.8.3)

There must be no damage to the main igniter chamber or adapter metal parts due to temperature effects.

Igniter insulation provided adequate thermal insulation.
(5.2)

ORIGINAL PAGE
BLACK AND WHITE PHOTOGRAPH

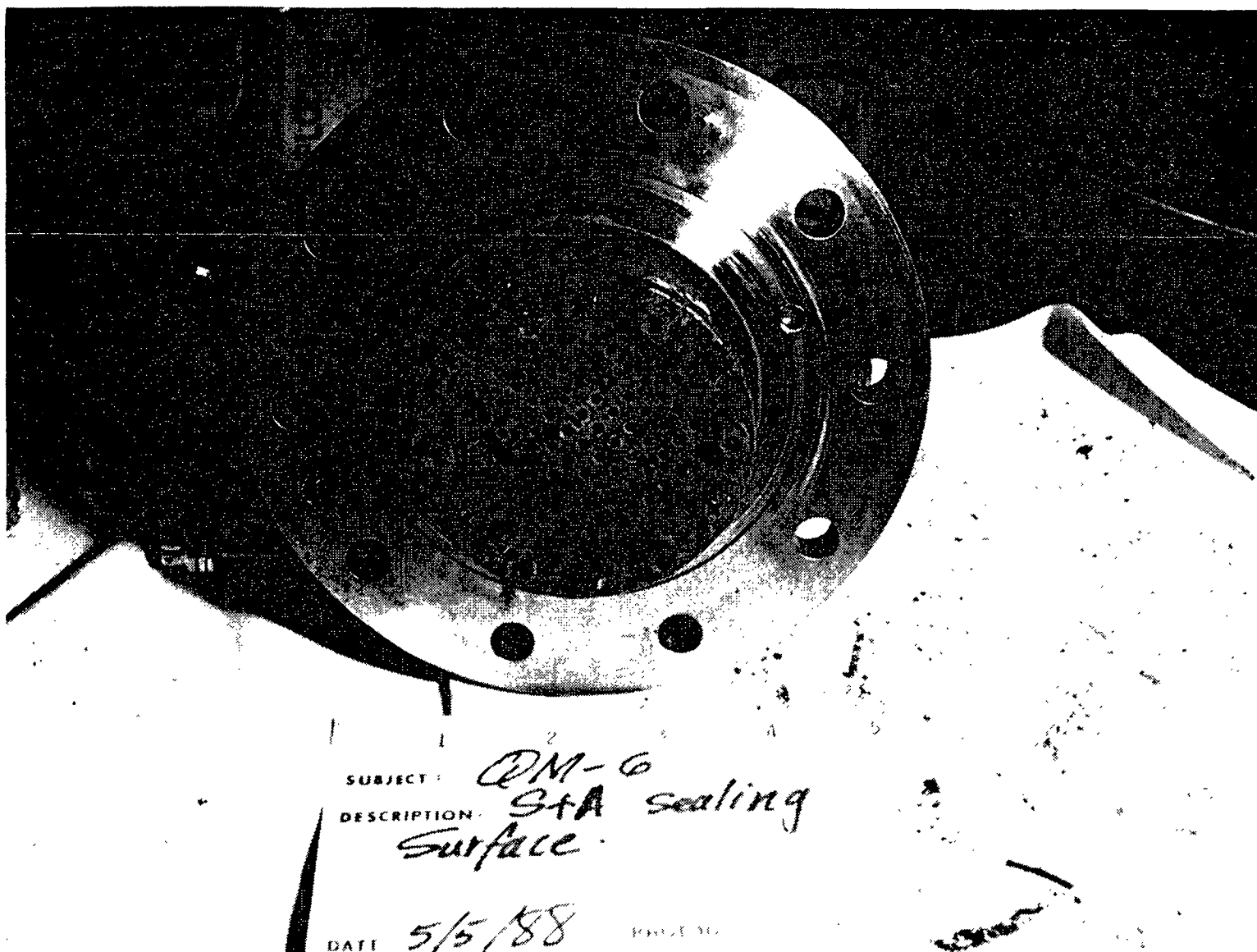


FIGURE 3. S&A SEALING SURFACE

SUBJECT:

DM-6

DESCRIPTION:

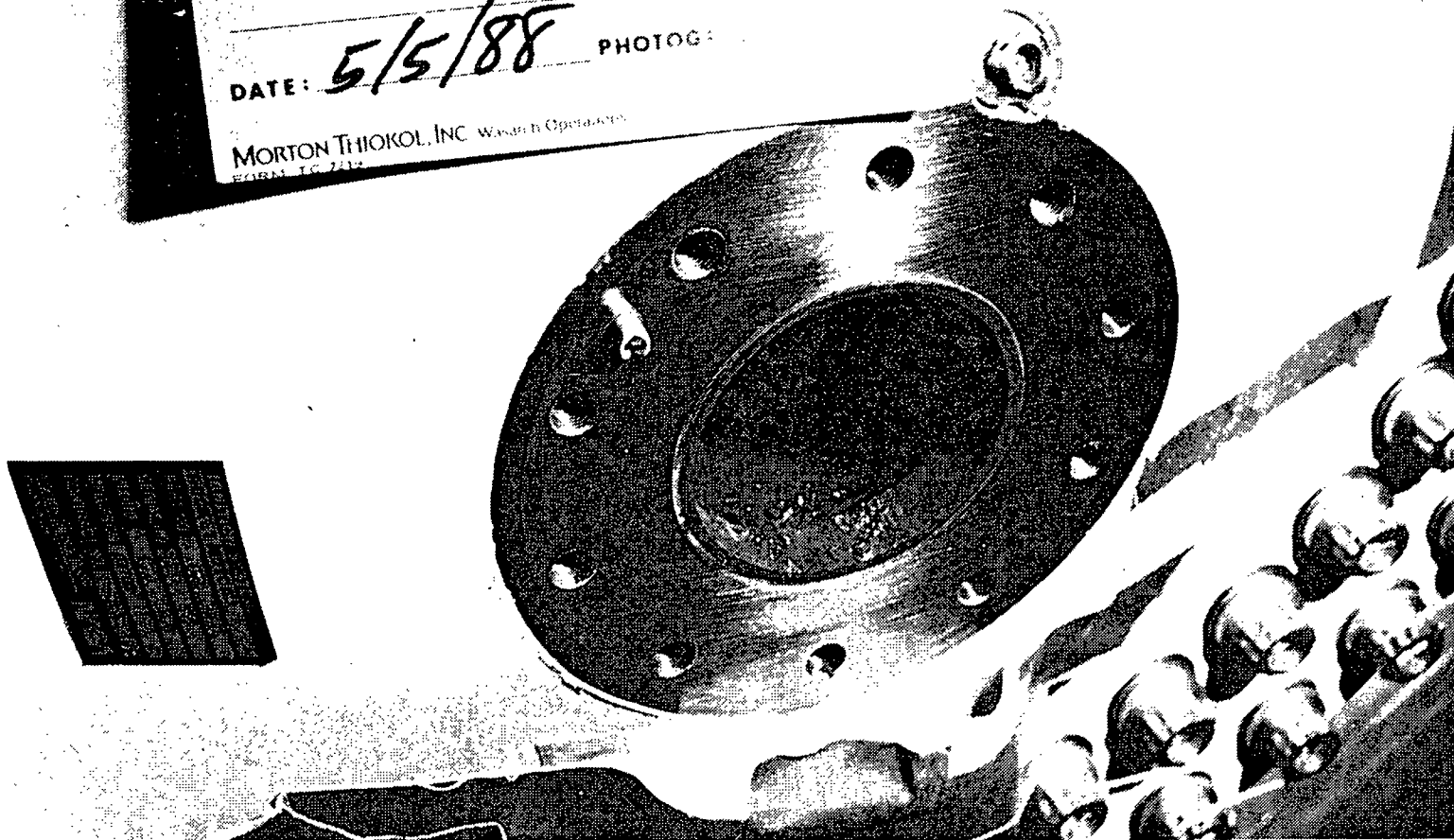
Adapter Sealing
Surface (S&A)

DATE:

5/5/88

PHOTO:

MORTON THIOKOL, INC. Wash. Operations
FORM TC 2-12



ORIGINAL PAGE
BLACK AND WHITE PHOTOGRAPH

FIGURE 4. S&A ADAPTER SEALING SURFACE

ORIGINAL PAGE
BLACK AND WHITE PHOTOGRAPH

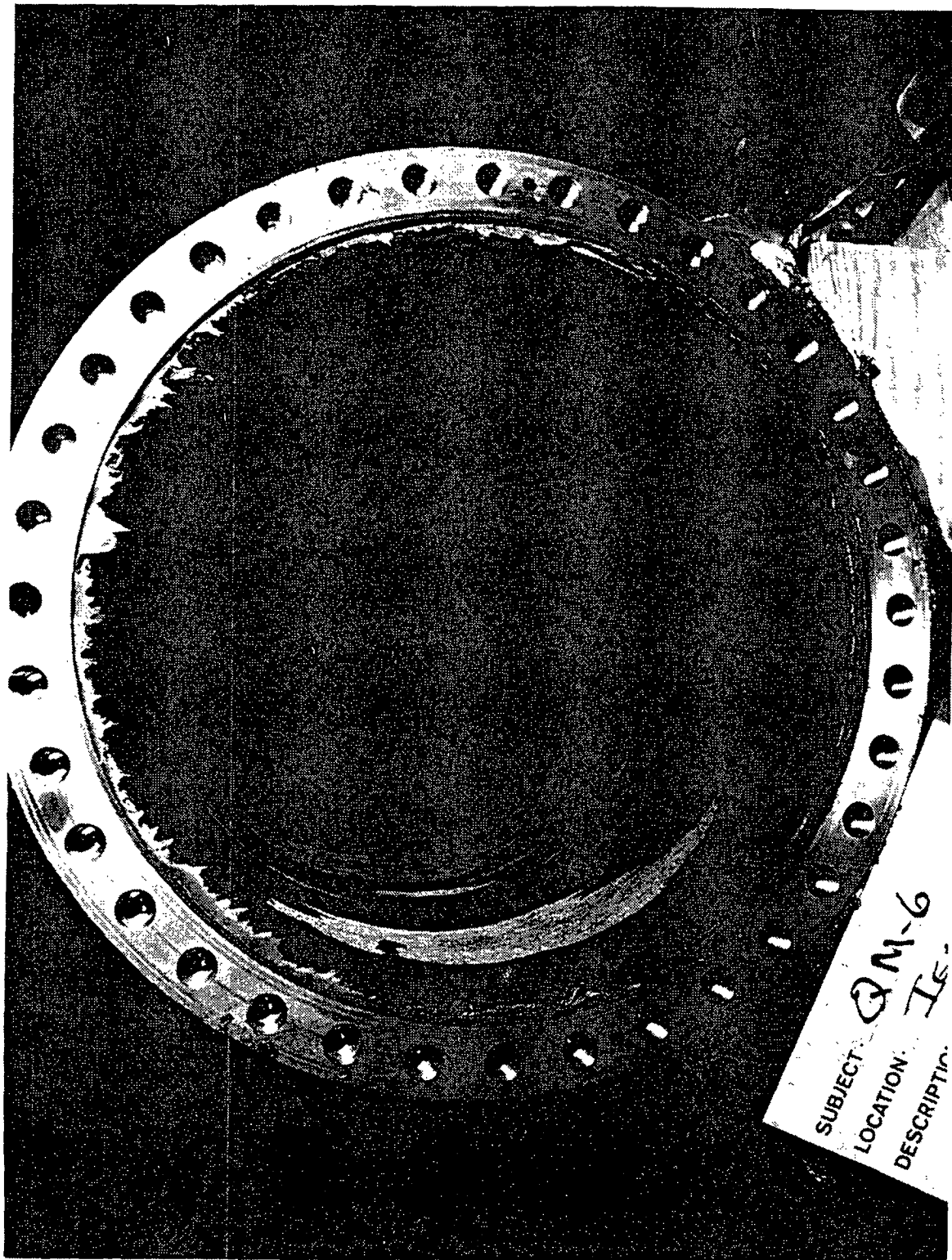


FIGURE 5. IGNITER CHAMBER

ORIGINAL PAGE
BLACK AND WHITE PHOTOGRAPH



FIGURE 6. IGNITER GASK-O-SEAL

ORIGINAL PAGE
BLACK AND WHITE PHOTOGRAPH

ORIGINAL PAGE IS
OF POOR QUALITY

QM-650870
IGNITER R.
P/N 107544-02-1901
3-15-88

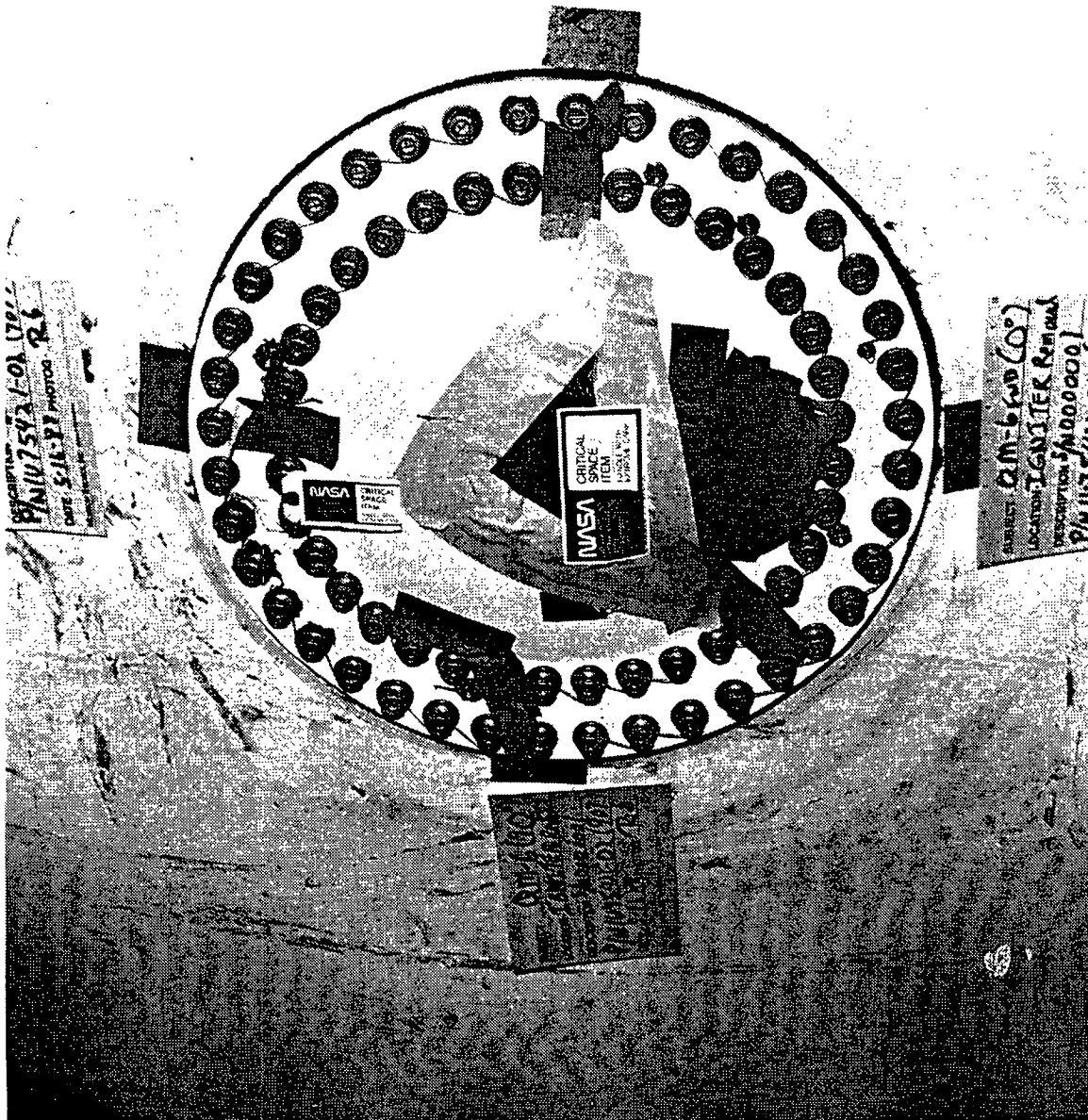


FIGURE 7. IGNITER PRIOR TO REMOVAL

ORIGINAL PAGE
BLACK AND WHITE PHOTOGRAPH

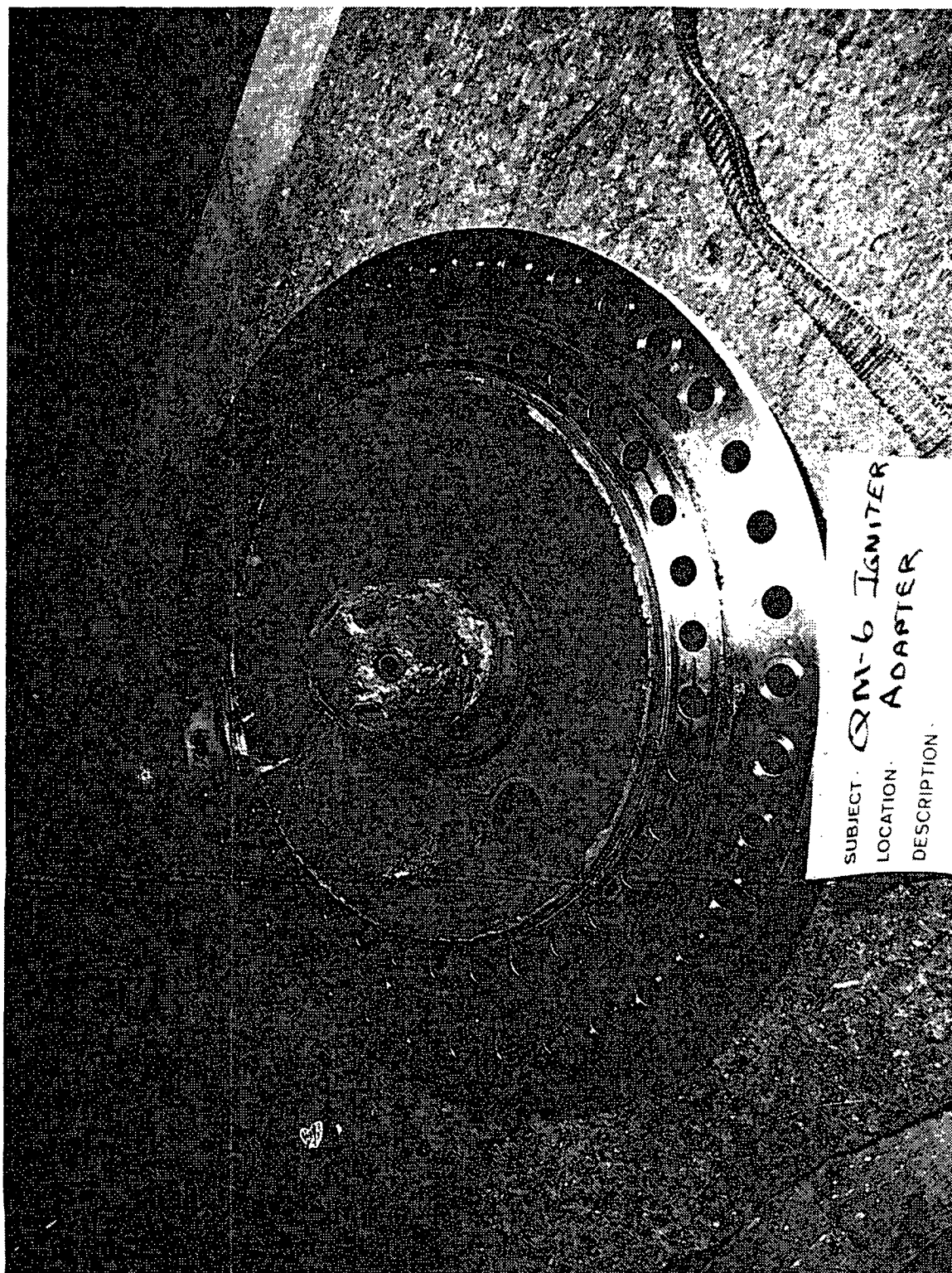


FIGURE 8. IGNITER ADAPTER

ORIGINAL PAGE
BLACK AND WHITE PHOTOGRAPH

ORIGINAL PAGE IS
OF POOR QUALITY

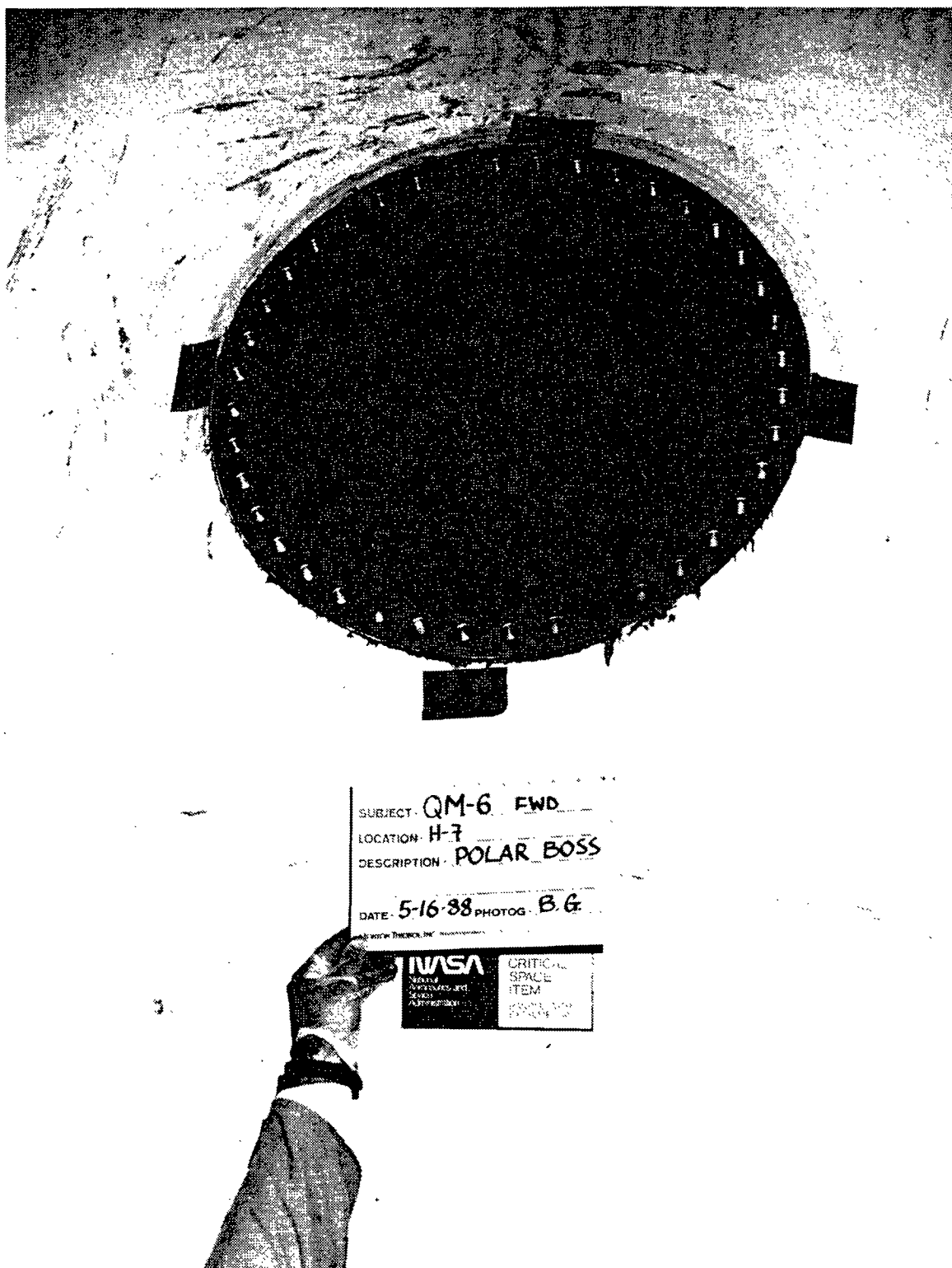


FIGURE 9. POLAR BOSS

ORIGINAL PAGE
BLACK AND WHITE PHOTOGRAPH

ORIGINAL PAGE IS
OF POOR QUALITY

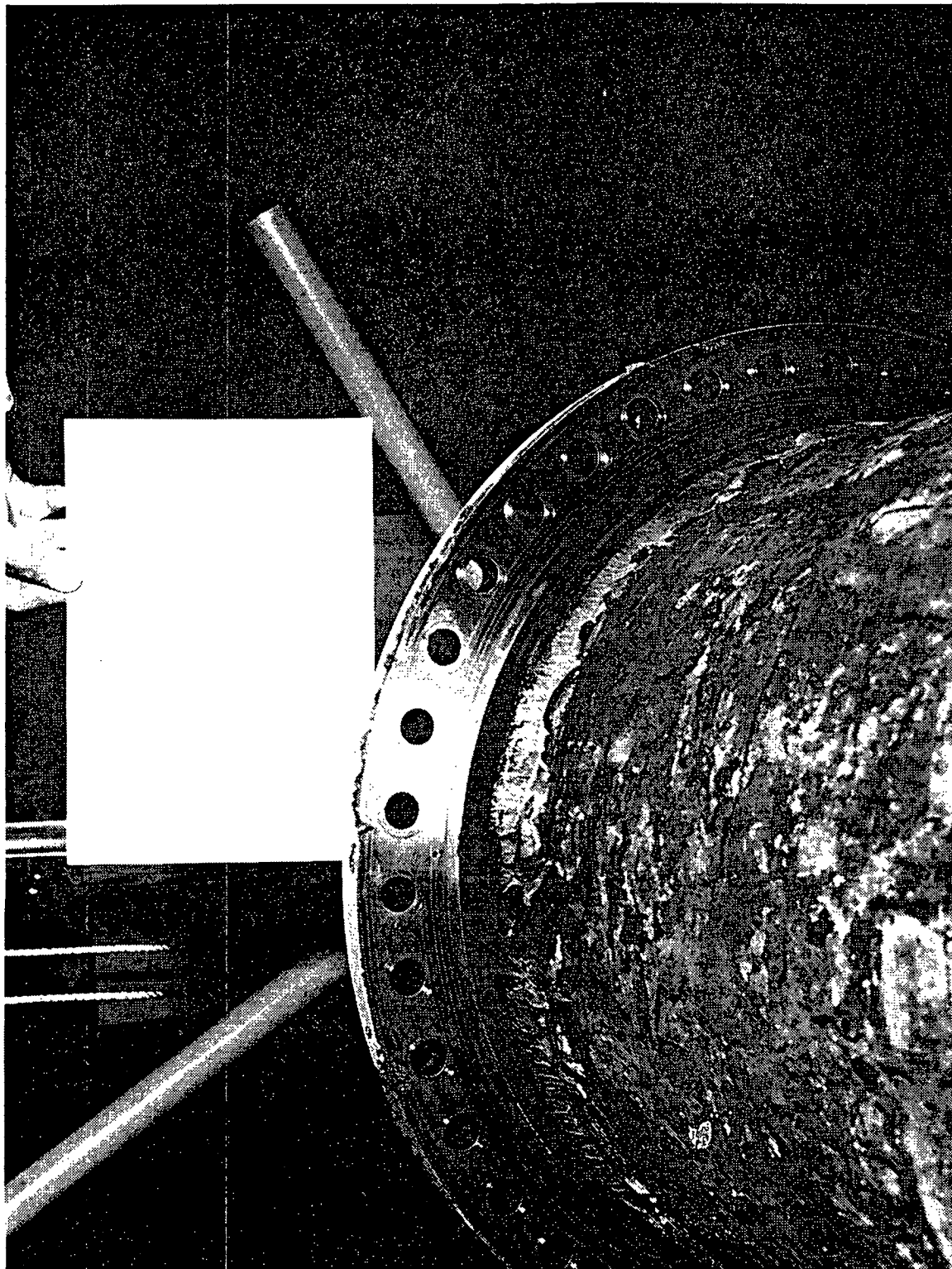


FIGURE 10. IGNITER ADAPTER SEALING SURFACE

ORIGINAL PAGE
BLACK AND WHITE PHOTOGRAPH

ORIGINAL PAGE IS
OF POOR QUALITY

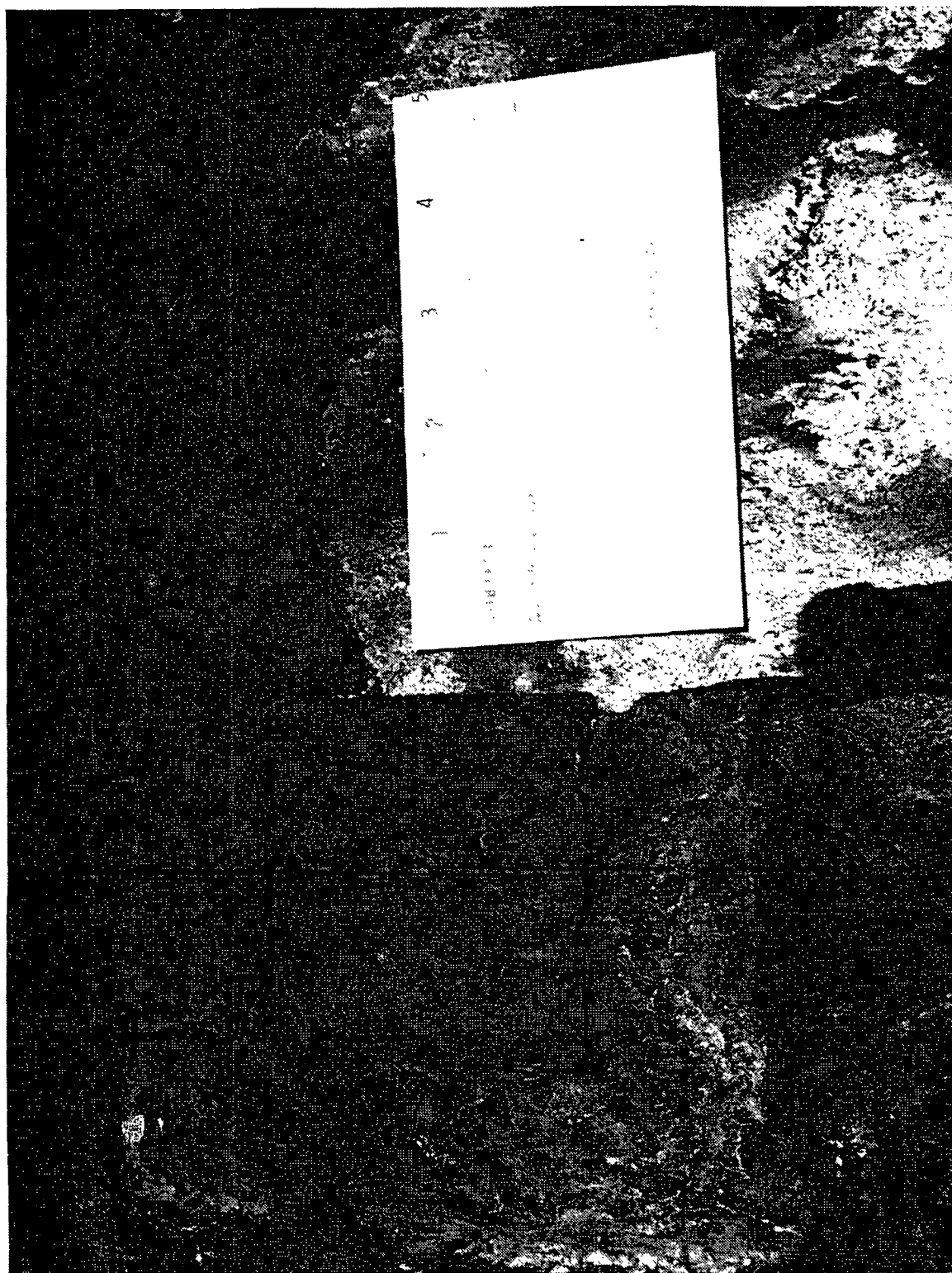
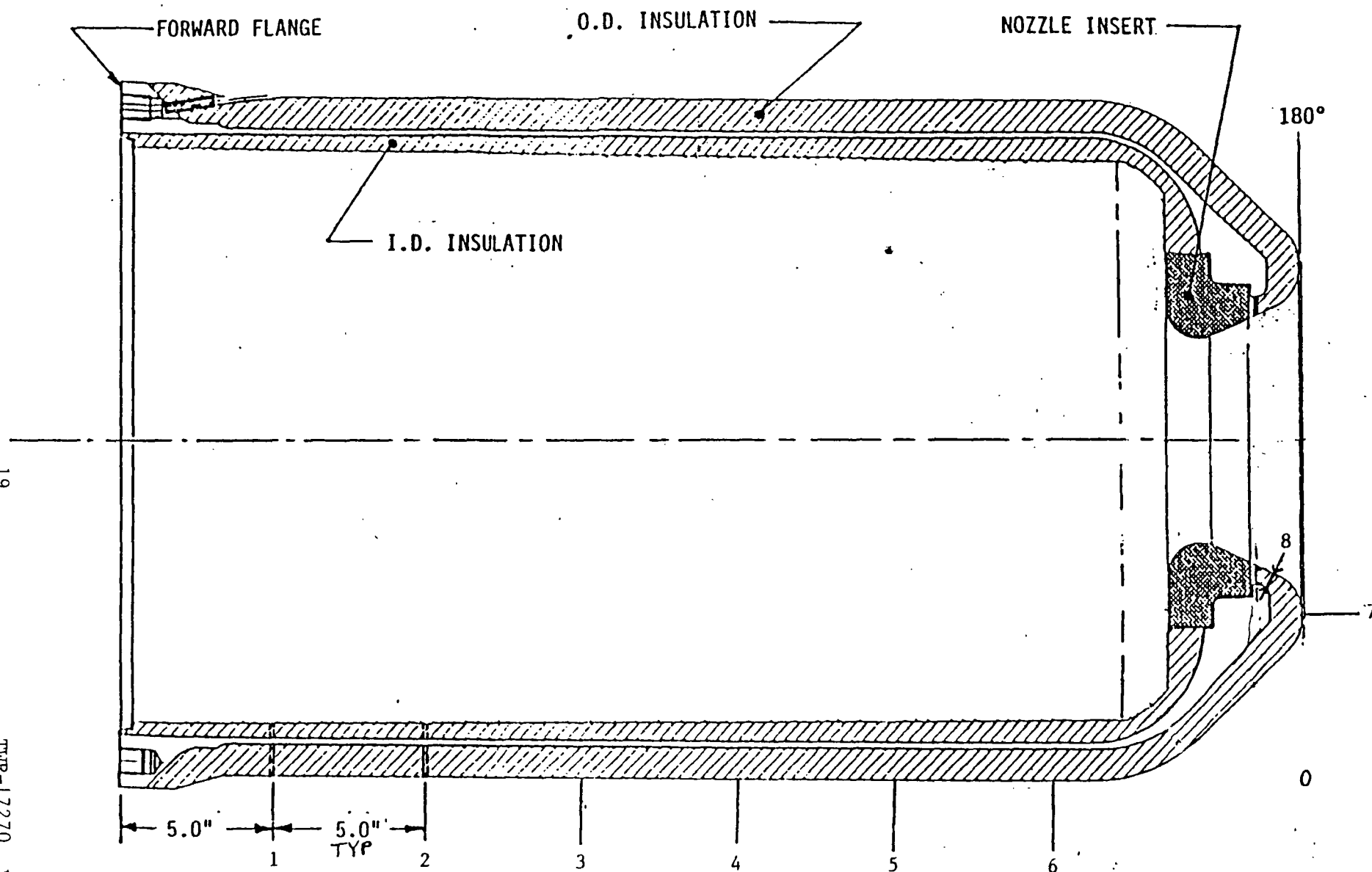


FIGURE 11. IGNITER INSULATION REPAIR PATCH



Post Test Plug Positioning For Insulation Measurements

FIGURE 12

IGNITER INSULATION HISTORY

MOTOR	LOCATION	PRE FIRE THICKNESS								POST FIRE THICKNESS								FACTOR OF SAFETY							
		1	2	3	4	5	6	7	8	1	2	3	4	5	6	7	8	1	2	3	4	5	6	7	8
DM-9	0 DEGREES	1.04	1.01	1	1	1.1	1.03	1.1	0.4	0.807	0.843	0.76	0.764	0.736	0.781	0.775	0.128	4.46	6.04	4.16	4.23	3.02	4.13	3.38	1.47
	180 DEGREES	0.99	1.01	1	0.98	1.02	1.04	1.07	0.4	0.872	0.816	0.787	0.796	0.762	0.731	0.761	0.142	8.38	5.20	4.69	5.32	3.95	3.36	3.46	1.55
QM-6	0 DEGREES	1.039	1.03	1.021	1.021	0.992	1.097	1.064	0.383	0.838	0.848	0.93	0.8	0.777	0.768	0.969	0.183	5.16	5.65	11.2	4.61	4.61	3.33	11.2	1.92
	180 DEGREES	0.99	0.967	1.009	1.011	1.006	1.026	1.003	0.382	0.799	0.82	0.838	0.738	0.759	0.722	0.96	0.182	5.18	5.91	5.90	3.70	4.07	3.37	23.3	1.91

N/A = NOT AVAILABLE

INCOR = MEASUREMENT TAKEN AT WRONG LOCATION OR
CHAR WAS NOT REMOVED

TABLE II. IGNITER INSULATION HISTORY

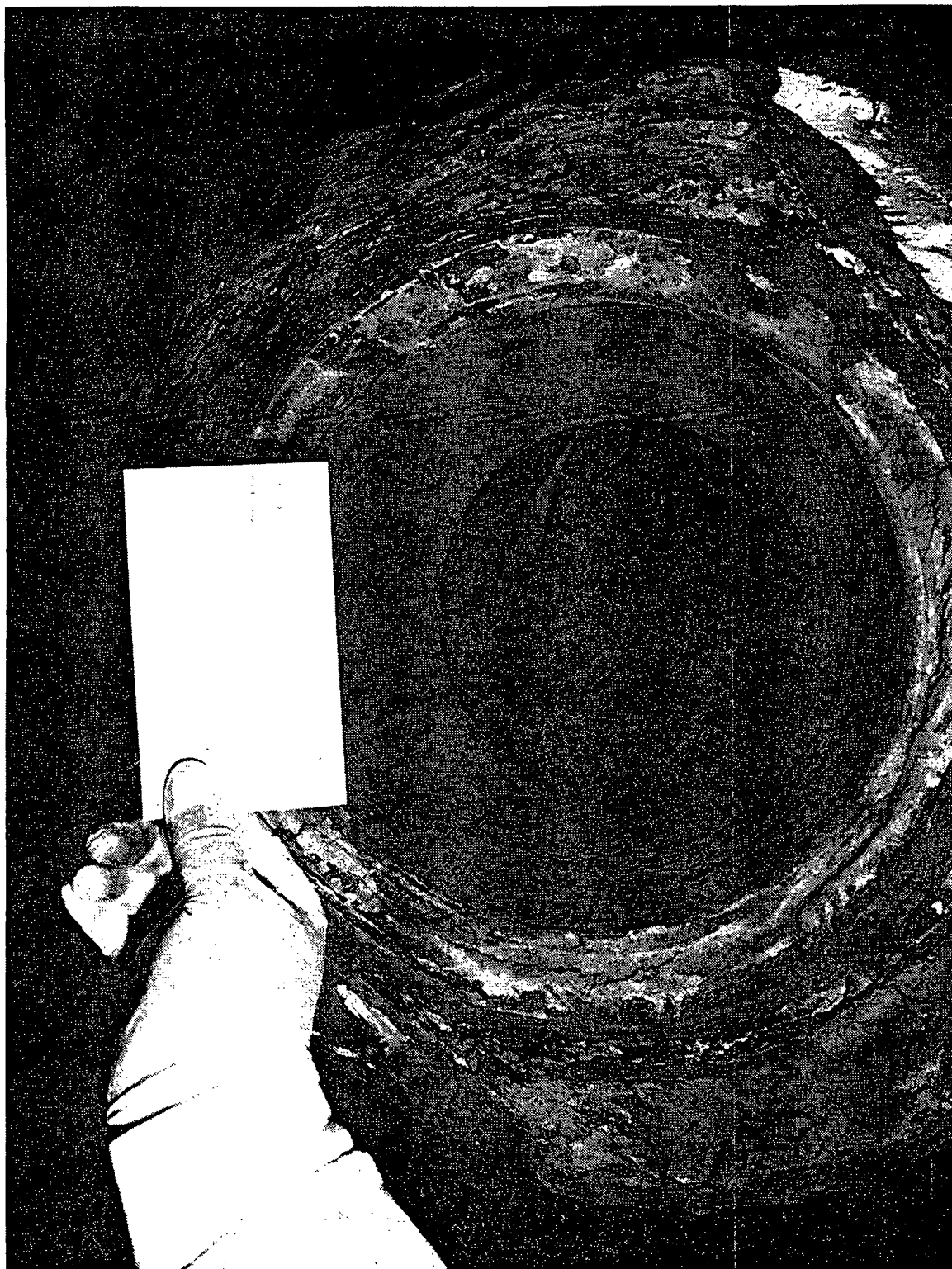


FIGURE 13. NOZZLE INSERT

ORIGINAL PAGE IS
OF POOR QUALITY

ORIGINAL PAGE
BLACK AND WHITE PHOTOGRAPH

nozzle throat measurements are as follow:

Position	Measurement
0 degrees	5.976 in.
60 degrees	5.980 in.
120 degrees	5.960 in.

6.0 METAL COMPONENTS

No structural damage or heat affect was noted on any metal components. Light corrosion was found on the adapter sealing surface. No bolts showed signs of yielding or damage. Figures 5 and 10 are pictures of the igniter case after disassembly (see page A-21).

7.0 IGNITION SYSTEM COMPONENT TEAM RECOMMENDATIONS

The Igniter System Components Program team has reviewed all observations and determined that only a potential anomaly exists. This is the blowhole found in the outer joint putty. The decision was made to continue to use the present design.

8.0 IGNITER PEDIGREE AND MEASURED PERFORMANCE (QM-6)

Part Number:	1U75163-01
Serial Number:	00000010
Lot Number:	38
Evaluation:	F700001
Casting Date:	11-Nov-1988
PBMT:	70° F
5 In CP Burn Rate:	.905 IN/SEC @ 2000 PSIA & 60° F
Target Burn Rate:	.877 IN/SEC @ 2000 PSIA & 60° F
Delivered Burn Rate:	.8789 IN/SEC @ 2000 PSIA & 60° F
Scale Factor:	.971
Igniter PMAX:	1903.8 PSIA
Time of IPGM:	.0900 SEC
Time of 10% PMAXIG:	.0238 SEC
Time of 700 PSIA:	.0332 SEC
Time of 90% PMAXIG:	.0532 SEC
Max Mass Flow Rate:	340.74 LBM/SEC
Throat Area:	28.51 M ²
C Star	5125.0

APPENDIX A
EVALUATION CHECKOFF WORKSHEETS

MORTON THIOKOL INC.

Wasatch Operations

Table A-II

Safe & Arm - Evaluation Checkoff Worksheet

Inspector(s): <u>D. Ballard</u>				
Motor No.: <u>QM-6</u>			Date: <u>5-10-88</u>	
I.	Heat Affect (Blisters or Discoloration)?	_____	yes	<u>X</u>
				no
II	Physical Damage (Nicks, Scratches, Gouges)?	_____	yes	<u>X</u>
				no
III.	Corrosion?	_____	yes	<u>X</u>
				no
If yes, note:				
	Condition (I, II, or III)	Degree Location (Deg.)	Circumferential Width (In.)	Degree Arc
	_____	_____	_____	_____
	_____	_____	_____	_____
	_____	_____	_____	_____
	_____	_____	_____	_____
Notes / Comments				

Table D-I
Preliminary Igniter Gasket - Evaluation Checkoff Worksheet

Inspector(s): <u>L. Nelsen, G. Abawi</u>				
Motor No.: <u>QM-6</u>			Date: <u>5/5/88</u>	
Gasket: ☐ Inner ☐ Outer ☒ S&A				
I. Gasket Identification Position <u>X</u> Forward Face <u>X</u> Aft Face				
II. Soot Past Seals? <u> </u> yes <u> </u> no				
If yes:				
Gasket Side	Seal I. D. (Prim. or Sec.)	Degree Location (Deg)	Degree Arc	
_____	_____	_____	_____	
_____	_____	_____	_____	
_____	_____	_____	_____	
_____	_____	_____	_____	
III. Foreign Material? <u> </u> yes <u>X</u> no				
If yes:				
Gasket Side	Material	Degree Location (Deg)	Degree Arc	
_____	_____	_____	_____	
_____	_____	_____	_____	
_____	_____	_____	_____	
_____	_____	_____	_____	
IV. Seal Damage? <u> </u> yes <u>X</u> no				
If yes:				
Gasket Side	Seal I. D. (Prim. or Sec.)	Condition	Degree Location (Deg)	Degree Arc
_____	_____	_____	_____	_____
_____	_____	_____	_____	_____
_____	_____	_____	_____	_____
_____	_____	_____	_____	_____
Notes / Comments				

Table D-I (Cont'd)

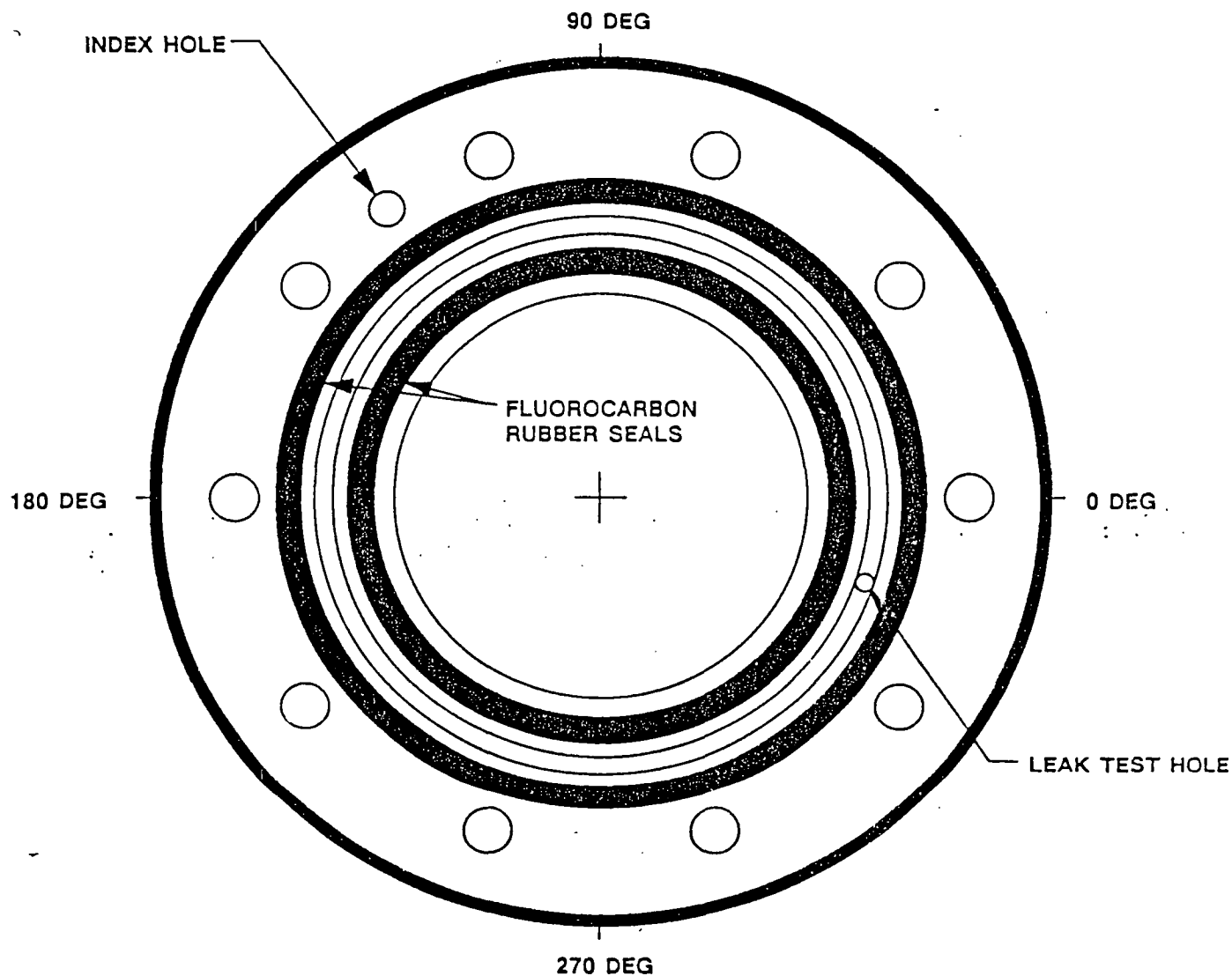
Preliminary Igniter Gasket - Evaluation Checkoff Worksheet

Inspector(s): <u>G. Abawi, L. Nelsen</u>			
Motor No.: <u>DM-6</u>		Date: <u>5/5/88</u>	
Gasket: ☐ Inner ☐ Outer ☒ S&A			
V. Heat Affected Seals or Retainer? ☐ yes ☒ no If yes, note the affected part (primary seal (PS), secondary seal (SS) or retainer (R)) and the indicated data:			
Affected Part (PS, SS, R)	Gasket Side	Degree Location (Deg)	Degree Arc
_____	_____	_____	_____
_____	_____	_____	_____
_____	_____	_____	_____
_____	_____	_____	_____
VI. Rust? ☐ yes ☒ no If yes:			
	Gasket Side	Degree Location (Deg)	Degree Arc
	_____	_____	_____
	_____	_____	_____
	_____	_____	_____
	_____	_____	_____
VII. Metal Damage? ☐ yes ☒ no If yes:			
Type of Damage	Gasket Side	Degree Location (Deg)	Degree Arc
_____	_____	_____	_____
_____	_____	_____	_____
_____	_____	_____	_____
_____	_____	_____	_____
Notes / Comments <u>Forward Face: A depression of approximately 0.25 in. Long, and 0.015 in deep was discovered on the crown of the primary seal of the S&A gasket.</u>			

MORTON THIOKOL INC.

Wasatch Operations

MOTOR NO. DM-6
INSPECTOR(S) L. Nelson
G. Abawi
DATE 5/5/88

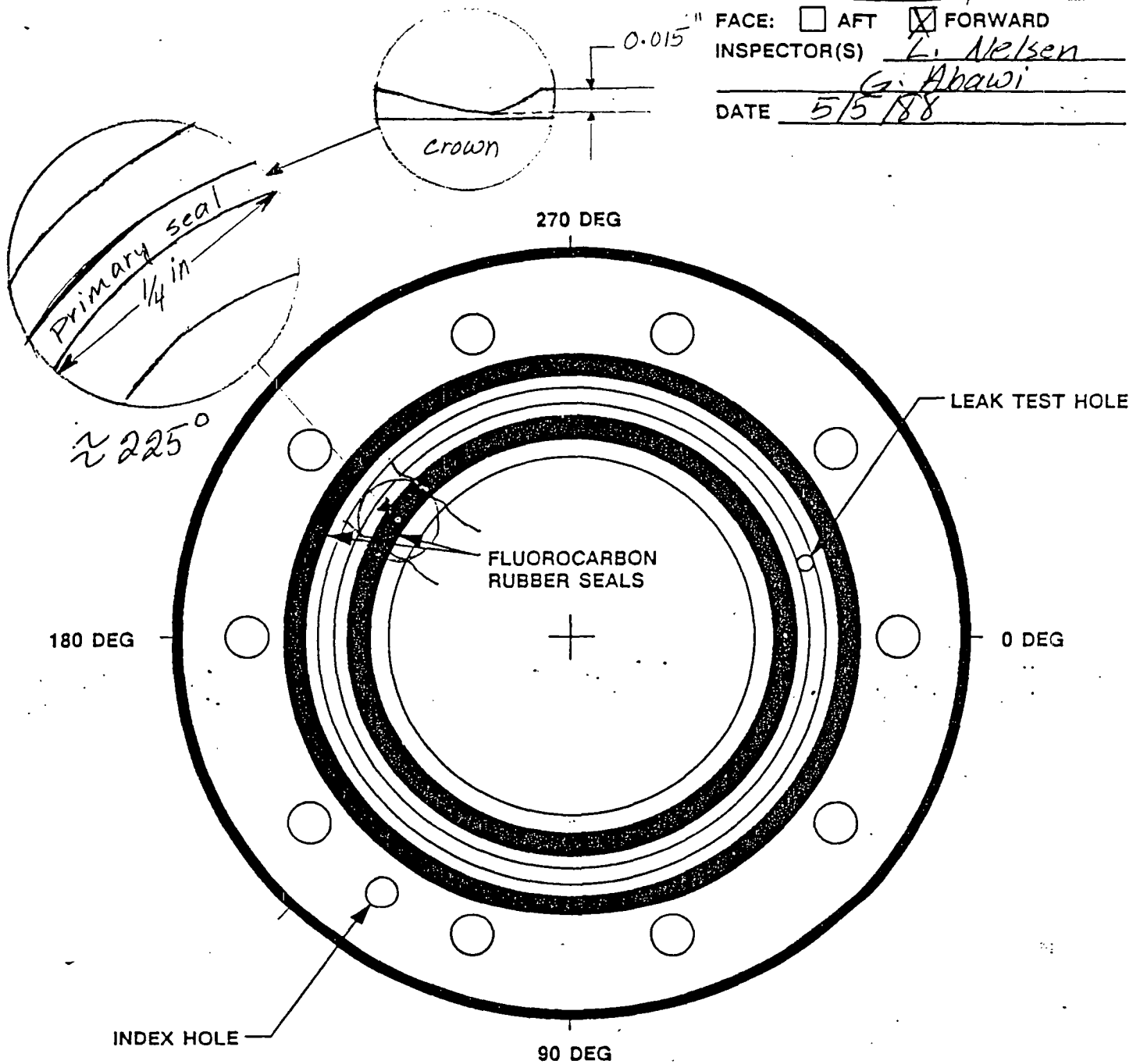


Observation Drawing Worksheet - Igniter S&A Gasket (Aft Face)
Figure D-2

REVISION _____

DOC NO.	TWR-17270	VOL	VI
SEC		PAGE	A4

MOTOR NO. QM-6
FACE: ☐ AFT ☒ FORWARD
INSPECTOR(S) L. Nelson
G. Abawi
DATE 5/5/88

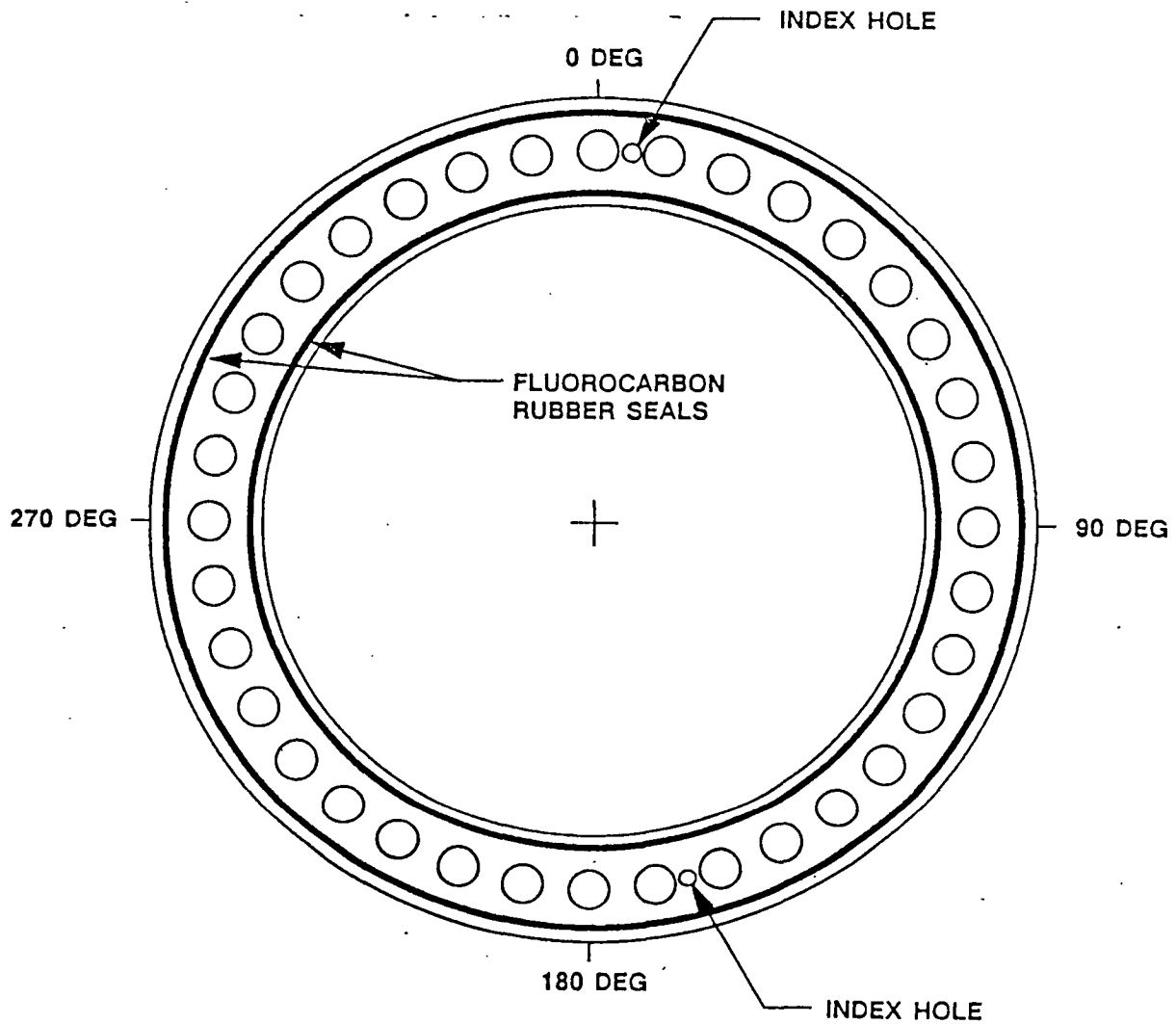


Observation Drawing Worksheet - Igniter S&A Gasket (Forward Face)
Figure D-1

REVISION _____

DOC NO.	TWR-17270	VOL	VI
SEC		PAGE	A5

MOTOR NO DM-6
INSPECTOR(S) G. Akkawi
DATE 6/3/88



Observation Drawing Worksheet - Igniter Inner Gasket (Aft Face)
Figure D-6

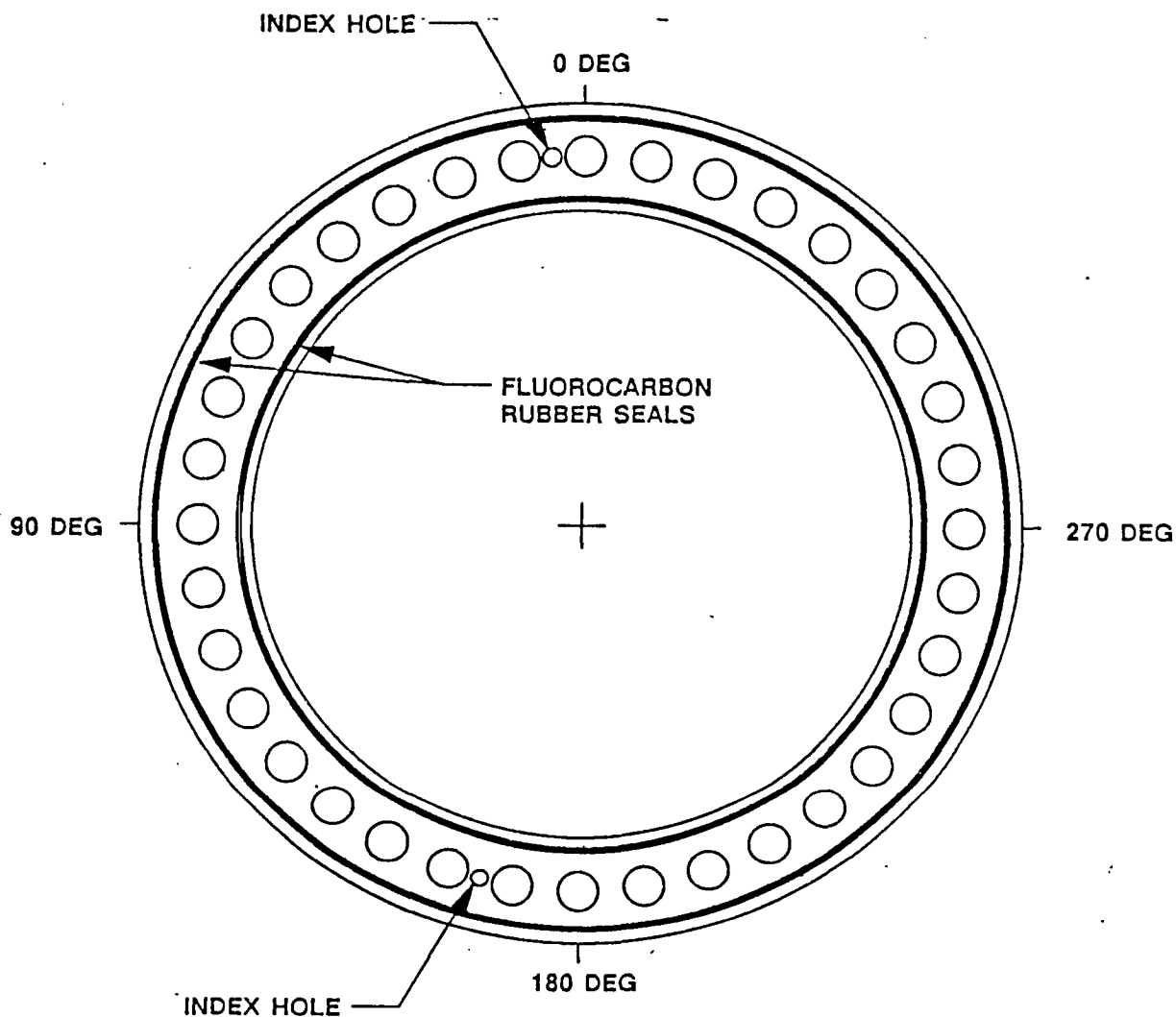
REVISION _____

DOC NO. TWR-17270
SEC. _____

PAGE _____

VOL VI

MOTOR NO QM-6
 INSPECTOR(S) G. Abawi
 DATE 6/3/88



Observation Drawing Worksheet - Igniter Inner Gasket (Forward Face)

Figure D-5

REVISION _____

DOC NO.	TWR-17270	VOL	VI
SEC		PAGE	A7

Table D-II

Detailed Igniter Gasket - Evaluation Checkoff Worksheet (A-2)

Inspector(s): <u>George Abawi</u>					
Motor No.: <u>QM-6</u>				Date: <u>6/3/88</u>	
Gasket: ☒ Inner ☐ Outer ☐ S&A					
I. Gasket Identification Position ☒ Forward Face ☒ Aft Face					
II. Soot Past Seals? ☐ yes ☒ no					
If yes:					
Gasket Side	Seal I. D. (Prim. or Sec.)	Degree Location (Deg)	Radial Distance (In.)	Circumferential Width (In.)	Degree Arc
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
III. Foreign Material? ☐ yes ☒ no					
If yes:					
Gasket Side	Material	Degree Location (Deg)	Circumferential Width (In.)	Degree Arc	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
IV. Seal Damage? ☐ yes ☒ no					
If yes:					
Gasket Side	Seal I. D. (Prim. or Sec.)	Condition	Degree Location (Deg)	Circumferential Width (In.)	Degree Arc
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
Notes / Comments					

Table D-II (Cont'd)

Detailed Igniter Gasket - Evaluation Checkoff Worksheet (A-2)

Inspector(s): <u>EA Akawi</u>					
Motor No.: <u>QM-6</u>				Date: <u>6/3/88</u>	
Gasket: ☒ Inner ☐ Outer ☐ S&A					
V. Heat Affected Seals or Retainer? <u> </u> yes <u> X </u> no If yes, note the affected part (primary seal (PS), secondary seal (SS) or retainer (R)) and the indicated data:					
Affected Part (PS, SS, R)	Gasket Side	Degree Location (Deg)	Circumferential Width (In.)	Degree Arc	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
VI. Rust? <u> </u> yes <u> X </u> no If yes:					
Gasket Side	Degree Location (Deg)	Radial Distance (In.)	Circumferential Width (In.)	Degree Arc	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
VII. Metal Damage? <u> </u> yes <u> X </u> no If yes:					
Type of Damage	Gasket Side	Degree Location (Deg)	Radial Distance (In.)	Circumferential Width (In.)	Degree Arc
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
Notes / Comments					

Wasatch Operations

Table D-III

Ignition System Sealing Surface Condition – Evaluation Checkoff Worksheet

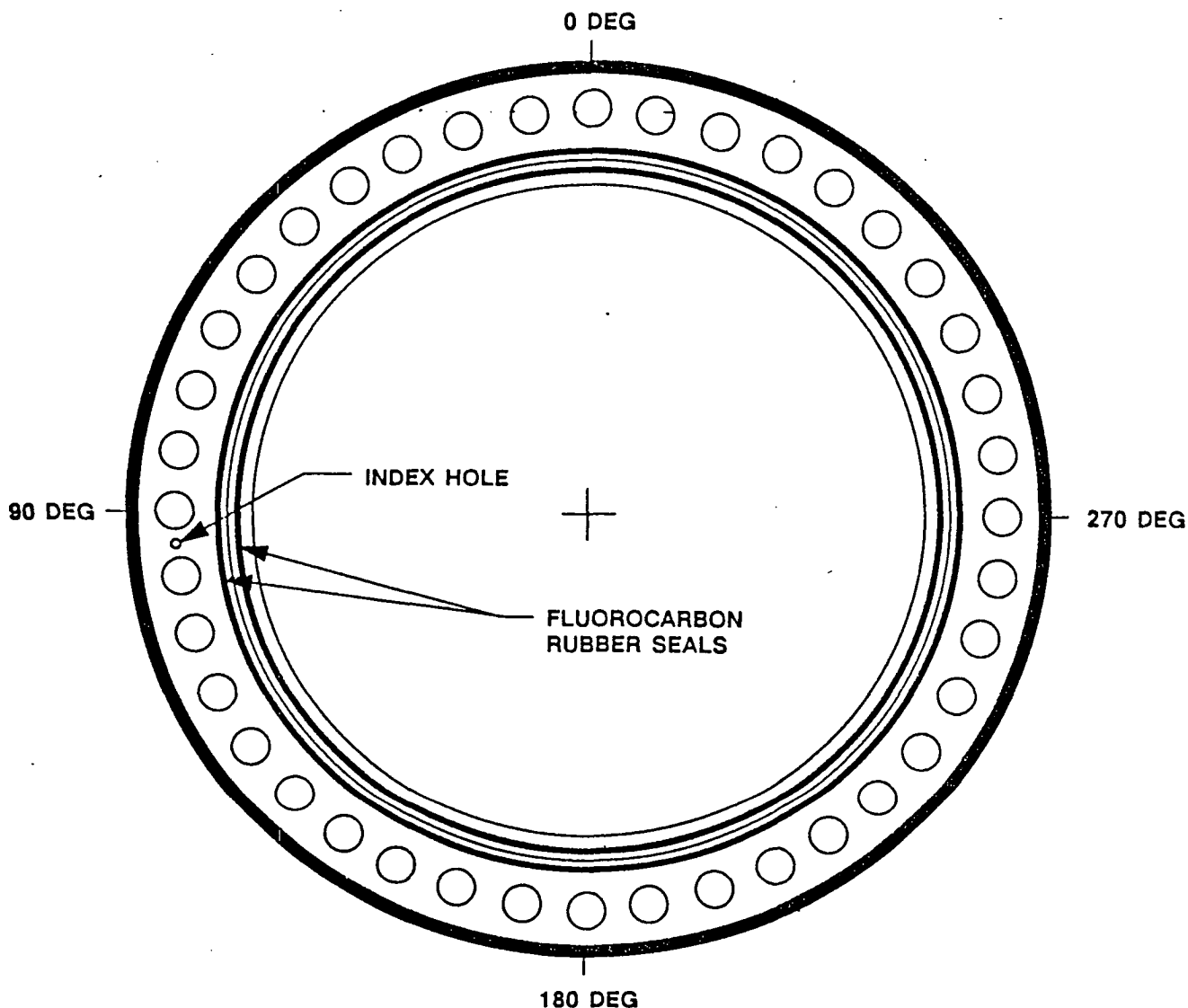
[illegible]

Table D-IV

Ignition System Joint Plug - Evaluation Checkoff Worksheet

Inspector(s): <u>George Abawi</u>			
Motor No.: <u>QM-6</u>		Date: <u>6/3/88</u>	
Joint: ☐ Adapter/Case ☒ Adapter/Chamber			
Degree Location: <u>305</u>			
I. Soot Past Seals?	_____ yes	X no	
II. Foreign Material?	_____ yes	X no	
III. Seal Damage?	_____ yes	X no	
IV. Heat Affected Seals or Plug?	_____ yes	X no	
V. Rust?	_____ yes	X no	
VI. Metal Damage?	_____ yes	X no	
If any of these conditions are found note the following:			
Condition (I, II, III, IV, V or VI)	Type of Material or Damage	Affected Part	Seal (Primary or Secondary)
_____	_____	_____	_____
_____	_____	_____	_____
_____	_____	_____	_____
_____	_____	_____	_____
Notes / Comments			

MOTOR NO. DM-6
INSPECTOR(S) G. Abuwi
DATE 5/16/88



Observation Drawing Worksheet - Igniter Outer Gasket (Forward Face)
Figure D-3

REVISION _____

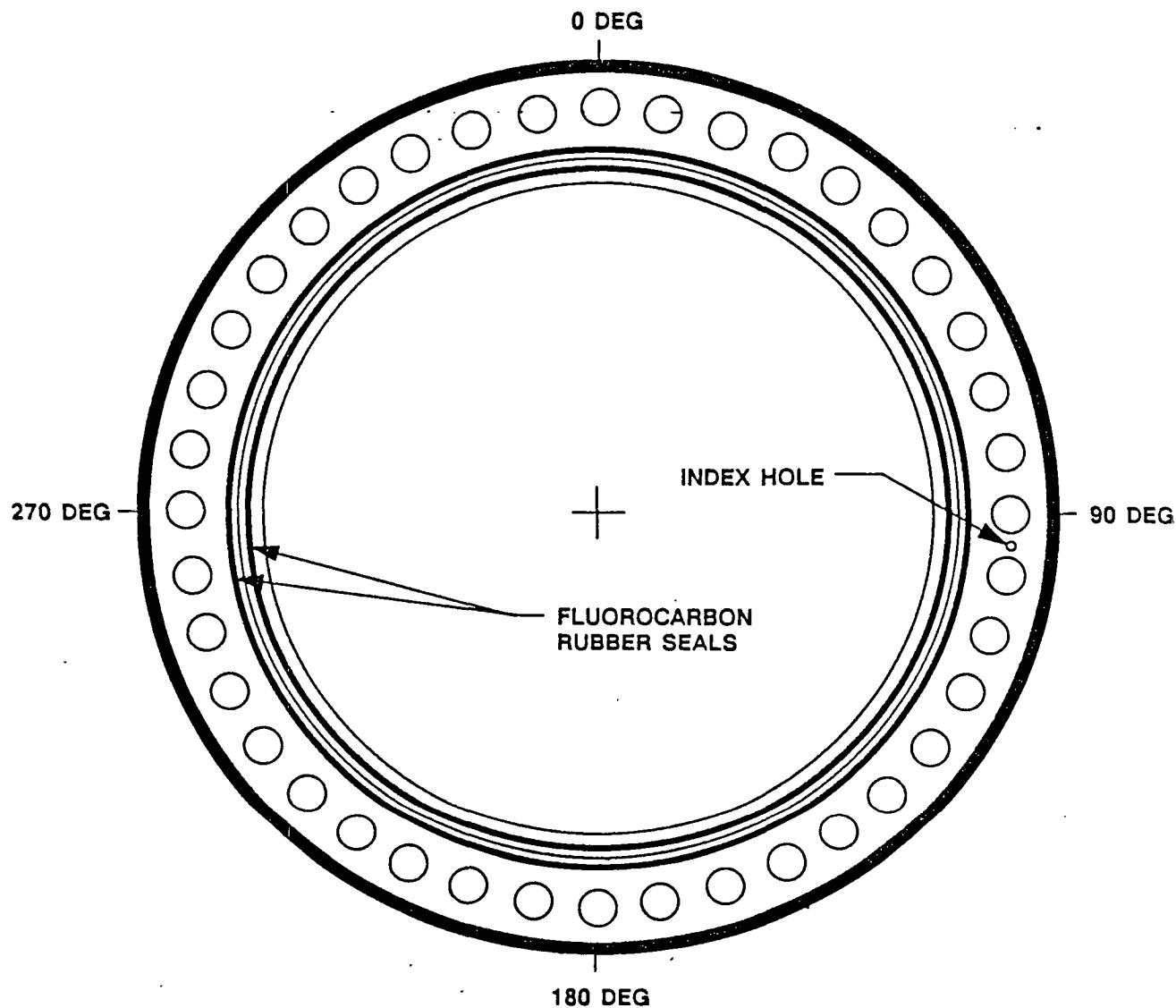
MORTON THIOKOL INC.

Wasatch Operations

MOTOR NO. DM-6

INSPECTOR(S) G. Abawi

DATE 5/16/88



Observation Drawing Worksheet - Igniter Outer Gasket (Aft Face)

Figure D-4

REVISION _____

DOC NO.	TWR-17270	VOL	VI
SEC		PAGE	A13

MURTON THIOKOL INC.

Wasatch Operations

Table D-II

Detailed Igniter Gasket - Evaluation Checkoff Worksheet (A-2)

Inspector(s): <u>G. Abawi</u>					
Motor No.: <u>QM-6</u>				Date: <u>5/16/88</u>	
Gasket: ☐ Inner ☒ Outer ☐ S&A					
I. Gasket Identification Position ☒ Forward Face ☒ Aft Face					
II. Soot Past Seals? ☐ yes ☒ no					
If yes:					
Gasket Side	Seal I. D. (Prim. or Sec.)	Degree Location (Deg)	Radial Distance (In.)	Circumferential Width (In.)	Degree Arc
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
III. Foreign Material? ☐ yes ☒ no					
If yes:					
Gasket Side	Material	Degree Location (Deg)	Circumferential Width (In.)	Degree Arc	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
IV. Seal Damage? ☐ yes ☒ no					
If yes:					
Gasket Side	Seal I. D. (Prim. or Sec.)	Condition	Degree Location (Deg)	Circumferential Width (In.)	Degree Arc
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
Notes / Comments					

Table D-II (Cont'd)

Detailed Igniter Gasket - Evaluation Checkoff Worksheet (A-2)

Inspector(s): <u>G. Abawi</u>					
Motor No.: <u>CM-6</u>				Date: <u>5/16/78</u>	
Gasket: ☐ Inner ☒ Outer ☐ S&A					
V. Heat Affected Seals or Retainer? <u> </u> yes <u>X</u> no If yes, note the affected part (primary seal (PS), secondary seal (SS) or retainer (R)) and the indicated data:					
Affected Part (PS, SS, R)	Gasket Side	Degree Location (Deg)	Circumferential Width (In.)	Degree Arc	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
VI. Rust? <u> </u> yes <u>X</u> no If yes:					
Gasket Side	Degree Location (Deg)	Radial Distance (In.)	Circumferential Width (In.)	Degree Arc	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
_____	_____	_____	_____	_____	
VII. Metal Damage? <u> </u> yes <u>X</u> no If yes:					
Type of Damage	Gasket Side	Degree Location (Deg)	Radial Distance (In.)	Circumferential Width (In.)	Degree Arc
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____
Notes / Comments					
<u>Scot on the inside edge of the gasket.</u>					

Wasatch Operations

Ignition System Sealing Surface Condition - Evaluation Checkoff Worksheet

REVISION _____

Table B-I
Igniter Insulation - Evaluation Checkoff Worksheet

Inspector(s): <u>P McCluskey</u>							
Motor No.: <u>QM-6</u>						Date: <u>5-16-88</u>	
Part: ☒ A. Adapter ☒ C. Initiator Chamber Exterior ☒ B. Igniter Chamber Exterior ☒ D. Igniter Chamber Interior							
I. Severe or Abnormal Insulation Erosion? <u> </u> yes <u>X</u> no II. Blistering? <u> </u> yes <u>X</u> no If yes, record:							
	Axial	Radial		Axial	Radial		
Condition	Location (In.)	Location (In.)	Degree	Length (In.)	Distance (In.)	Circumferential	Degree
(I or II)	(B, C, D Only)	(A Only)	Location (Deg.)	(B,C,D Only)	(A Only)	Width (In.)	Arc
_____	_____	_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____	_____	_____
_____	_____	_____	_____	_____	_____	_____	_____

Notes / Comments

- 1) Insulation char and erosion looked normal. Thickness checks will be taken later
- 2) The repair patch at 0° appeared to have performed as expected. ~~the~~ Patch was eroded away but the plug remained in place. After chamber insulation is washed out further inspection will be performed
- 3) Three holes were repaired at the aft end of the igniter. There was no evidence ~~of the~~ repair failed. Detailed inspection will be done ~~at~~ after insulation removal.

 ORIGINAL PAGE IS
OF POOR QUALITY

MORTON THIOKOL INC.

Wasatch Operations

Table A-I

External Ignition System - Evaluation Checkoff Worksheet

Inspector(s): <u>P McCluskey</u>																																									
Motor No.: <u>QM-6</u>					Date: <u>5-16-88</u>																																				
I. Hotspots (Blisters or Discoloration) on Igniter Adapter? <u> </u> yes <u> X </u> no If yes: <table style="width:100%; border-collapse: collapse;"> <thead> <tr> <th style="width:15%;">Condition</th> <th style="width:15%;">Degree Location (Deg.)</th> <th style="width:15%;">Radial Location (In.)</th> <th style="width:15%;">Radial Distance (In.)</th> <th style="width:15%;">Circumferential Width (In.)</th> <th style="width:15%;">Degree Arc</th> </tr> </thead> <tbody> <tr><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td></tr> </tbody> </table>							Condition	Degree Location (Deg.)	Radial Location (In.)	Radial Distance (In.)	Circumferential Width (In.)	Degree Arc																													
Condition	Degree Location (Deg.)	Radial Location (In.)	Radial Distance (In.)	Circumferential Width (In.)	Degree Arc																																				
II. Physical Damage (Nicks, Scratches, Gouges)? A. Adapter <u> </u> yes <u> X </u> no B. Adapter Bolts (Outer Circle) <u> </u> yes <u> X </u> no C. Adapter Bolts (Inner Circle) <u> </u> yes <u> X </u> no D. S&A Bolts <u> </u> yes <u> X </u> no If yes, note the affected part (A, B, C or D) and the indicated data: <table style="width:100%; border-collapse: collapse;"> <thead> <tr> <th style="width:10%;">Affected Part</th> <th style="width:15%;">Condition</th> <th style="width:15%;">Degree Location (Deg.)</th> <th style="width:15%;">Radial Location (In.)</th> <th style="width:15%;">Radial Distance (In.)</th> <th style="width:15%;">Circumferential Width (In.)</th> <th style="width:15%;">Degree Arc</th> </tr> </thead> <tbody> <tr><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td></tr> </tbody> </table>							Affected Part	Condition	Degree Location (Deg.)	Radial Location (In.)	Radial Distance (In.)	Circumferential Width (In.)	Degree Arc																												
Affected Part	Condition	Degree Location (Deg.)	Radial Location (In.)	Radial Distance (In.)	Circumferential Width (In.)	Degree Arc																																			
III. Corrosion? A. Adapter <u> </u> yes <u> X </u> no B. Adapter Bolts (Outer Circle) <u> </u> yes <u> X </u> no C. Adapter Bolts (Inner Circle) <u> </u> yes <u> X </u> no D. S&A Bolts <u> </u> yes <u> X </u> no If yes, note the affected part (A, B, C or D) and the indicated data: <table style="width:100%; border-collapse: collapse;"> <thead> <tr> <th style="width:10%;">Affected Part</th> <th style="width:15%;">Degree Location (Deg.)</th> <th style="width:15%;">Radial Location (In.)</th> <th style="width:15%;">Radial Distance (In.)</th> <th style="width:15%;">Circumferential Width (In.)</th> <th style="width:15%;">Degree Arc</th> </tr> </thead> <tbody> <tr><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td></tr> <tr><td> </td><td> </td><td> </td><td> </td><td> </td><td> </td></tr> </tbody> </table>							Affected Part	Degree Location (Deg.)	Radial Location (In.)	Radial Distance (In.)	Circumferential Width (In.)	Degree Arc																													
Affected Part	Degree Location (Deg.)	Radial Location (In.)	Radial Distance (In.)	Circumferential Width (In.)	Degree Arc																																				
Notes / Comments																																									

Table A-V

Igniter Putty Condition - Evaluation Checkoff Worksheet

335°

350°

adapter

gaskets

chamber

1.00

Insulation

MORTON THIOKOL INC.

Wasatch Operations

Table A-VI
Igniter Nozzle Insert - Evaluation Checkoff Worksheet

Inspector(s): <u>P McCluskey</u>				
Motor No.: <u>QM-C</u>			Date: <u>5-16-88</u>	
I. Cracked Nozzle Insert?	<u> </u> yes		<u>X</u> no	
II. Chipped Nozzle Insert?	<u> </u> yes		<u>X</u> no	
If Yes:				
Condition (I or II)	Degree Location (Deg.)	Axial Length (In.)	Circumferential Width (In.)	Degree Arc
<u> </u>	<u> </u>	<u> </u>	<u> </u>	<u> </u>
<u> </u>	<u> </u>	<u> </u>	<u> </u>	<u> </u>
<u> </u>	<u> </u>	<u> </u>	<u> </u>	<u> </u>
<u> </u>	<u> </u>	<u> </u>	<u> </u>	<u> </u>
Notes / Comments <u>cracks in the char layer. NO</u> <u>apparent damage to the insert.</u>				

Wasatch Operations.

Ignition System Sealing Surface Condition - Evaluation Checkoff Worksheet

REVISION_____