
ADVANCED SOLID ROCKET MOTOR PROJECT STATUS

Tenth Workshop for Computational Fluid Dynamics (CFD) Applications in Rocket Propulsion

28 April 1992

Keith Coates
EE 71
MSFC

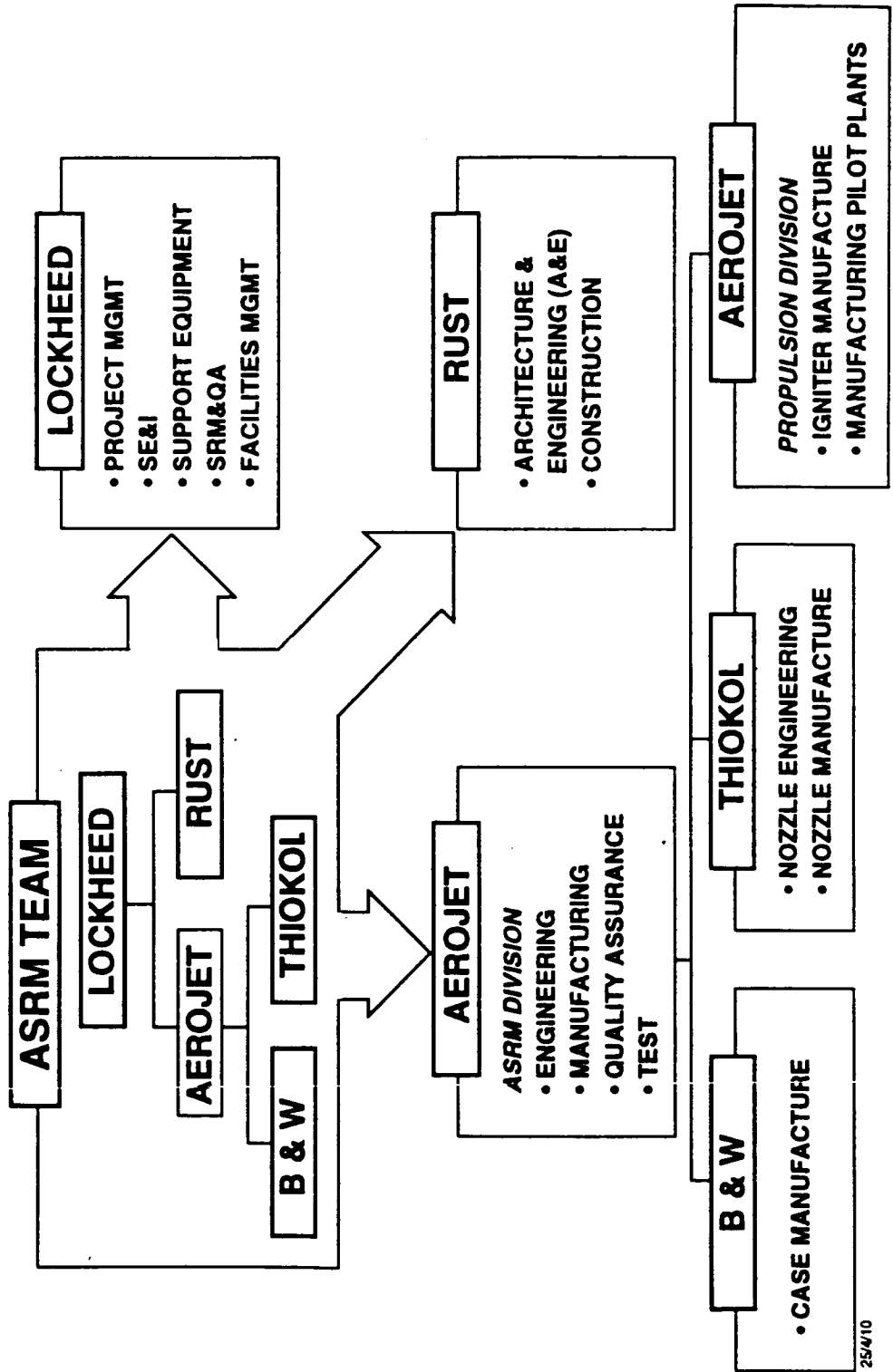
OUTLINE

- **Project Objectives**
- **Team**
- **Locations**
- **Motor Design**
- **Schedule**
- **Technical Issues**

PROJECT OBJECTIVES

- **Improve System Safety and Reliability**
 - Design Features
 - Enhanced Quality
 - Reproducibility
- **Improve Shuttle Payload Performance: 12,000 lb**
- **Optimize Program Cost**
- **Promote Competitive SRM Industry**
 - Construct and Operate Government Owned Manufacturing and Test Installations

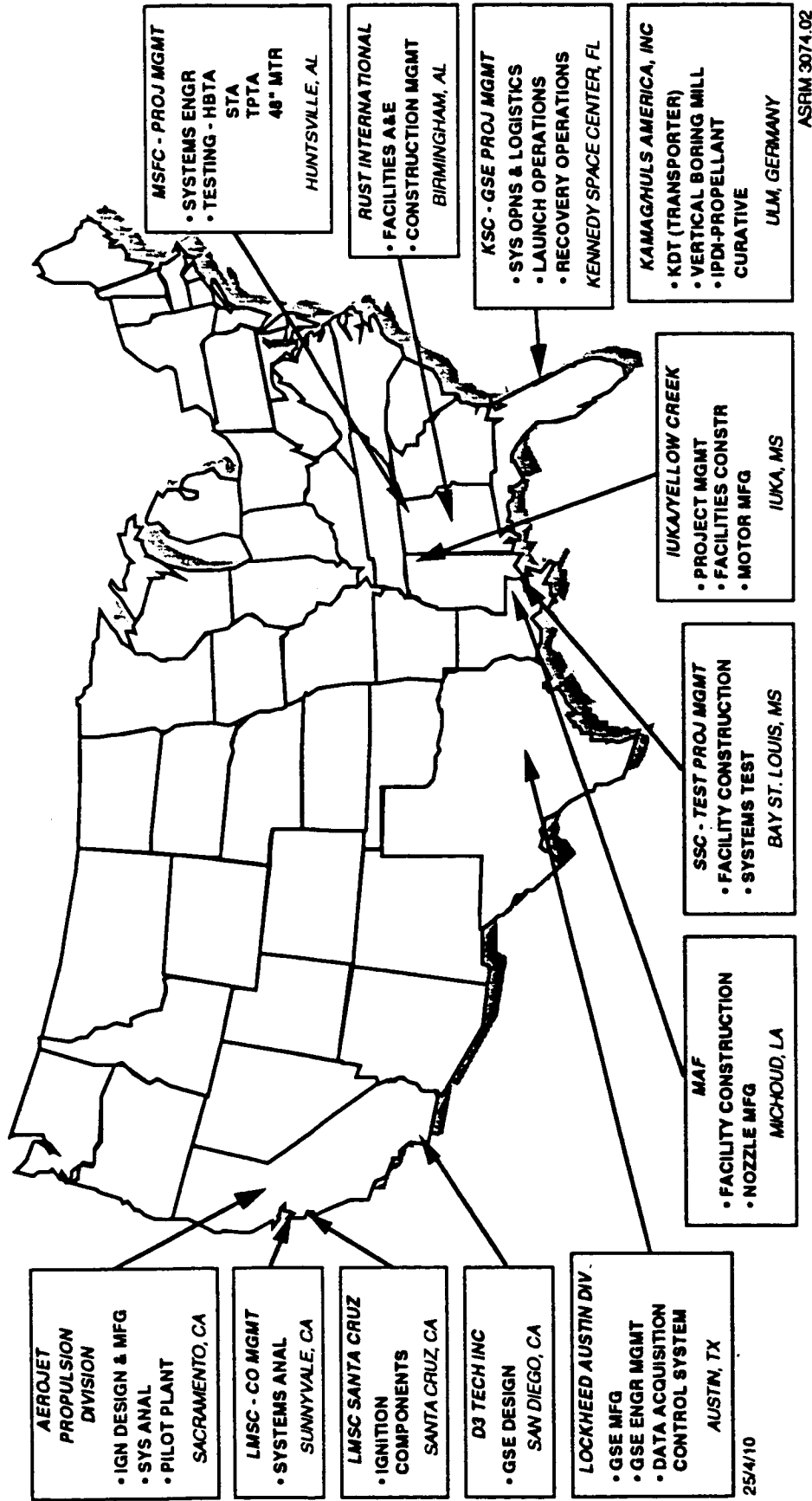
ASRM PROJECT TEAM



ASRM 3074.01

25/4/10

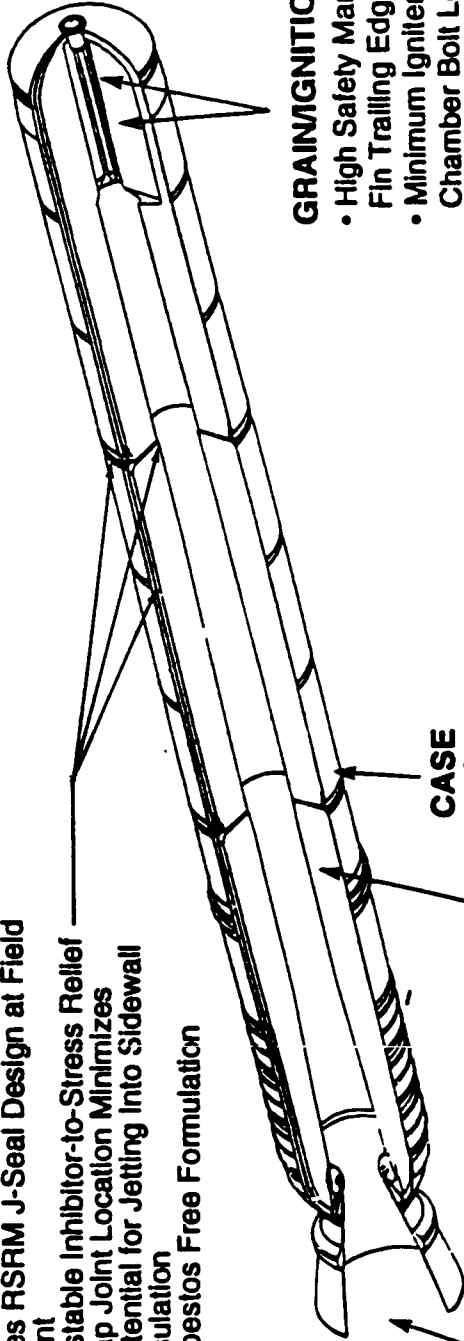
ASRM PROJECT TEAM LOCATIONS



ASRM MOTOR DESIGN HIGHLIGHTS

INSULATION

- Uses RSRM J-Seal Design at Field Joint
- Castable Inhibitor-to-Stress Relief Flap Joint Location Minimizes Potential for Jetting into Sidewall Insulation
- Asbestos Free Formulation



NOZZLE

- Minimum Joints and Inlet/throat rings
- Improved process ablative materials
- Eliminates flexseal cowl and boot assembly

CASE

- Two Field Joints
- High Fracture Toughness
- High Stress Corrosion Resistance
- Welded Factory Joints
- Integral Stiffeners and ET Attach Ring in Aft Segment Eliminates Failure Points Experienced With Bolt-On Stiffeners and Ring

PROPELLANT

- Industry Proven HTPB Propellant with over 60M lbs Successfully Produced
- Formulated and Proven for Continuous Mix Process
- Positive Margins

GRAINIGNITION

- High Safety Margin in Forward Fin Trailing Edge
- Minimum Igniter Chamber Bolt Leak Paths
- Expendable Carbon Filament Chamber
- TBI Initiator Eliminates S&A Device Leak Paths

ASRM DESIGN PARAMETERS

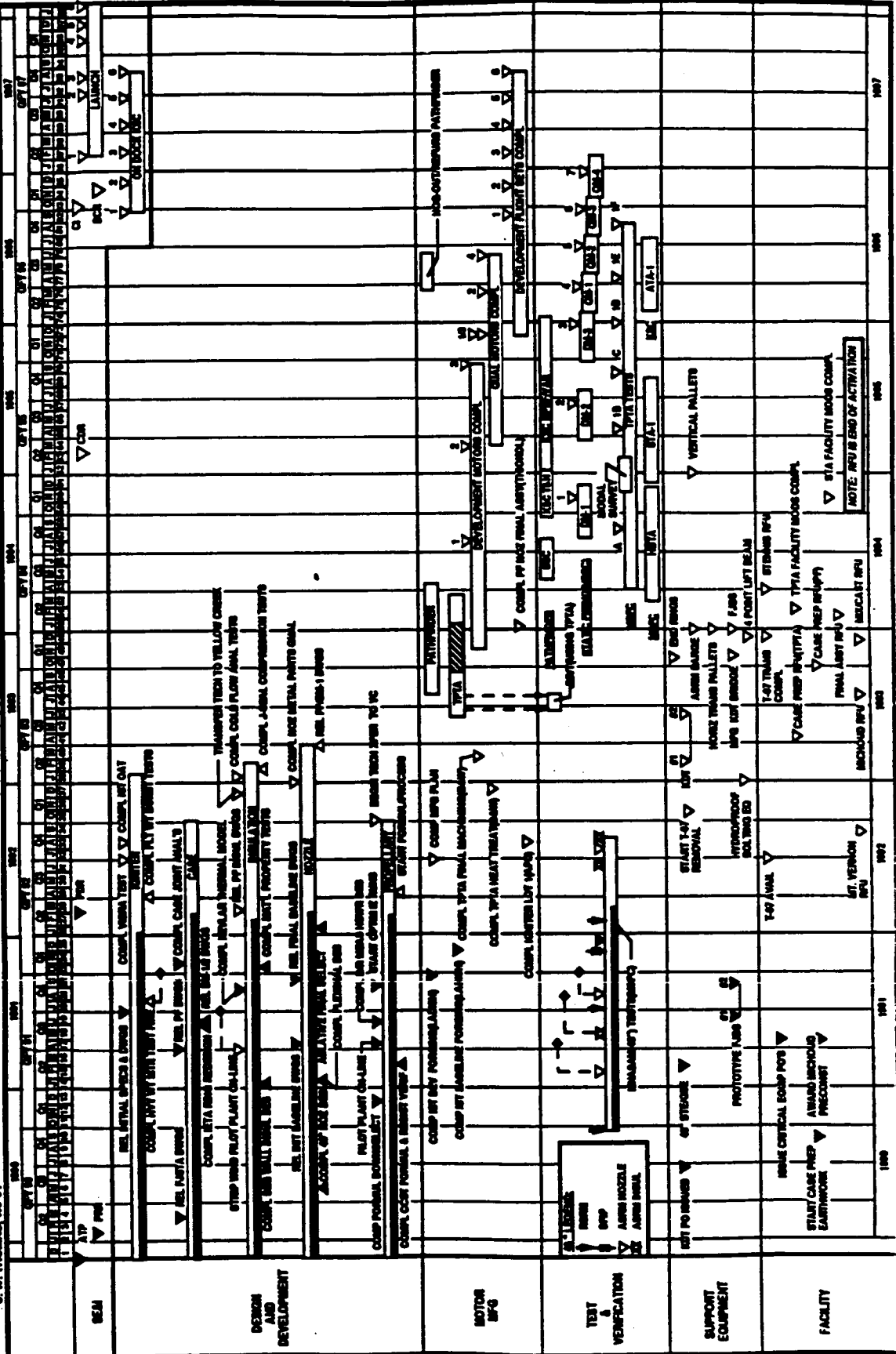
■ Diameter/Length, in	150/1,513.43
■ Average Thrust Vacuum, lbf; Web time	2,654,776
■ Delivered Isp Vacuum, sec	268.1
■ Area Ratio, (Ae/At)	7.48
■ Motor Weight, lb	1,351,092
■ Propellant Weight, lb	1,209,589
■ Motor Propellant Mass Fraction, (Wp/Wt)	0.895
■ Inert Weight , lb	141,503
<input type="checkbox"/> Metal Case Weight/Number of Segments, lb	98,553/3
<input type="checkbox"/> Single Nozzle Weight, lb	18,800
■ Solid Propellant Type	HTPB
■ Average Chamber Pressure, psia; Action Time	612
■ Burn Rate at 625 psia, in/sec	0.350
■ Action Time, sec	130.9
■ Thrust Vector Control	Flexible Bearing
■ Recovery/Reuse	Yes

Project Status

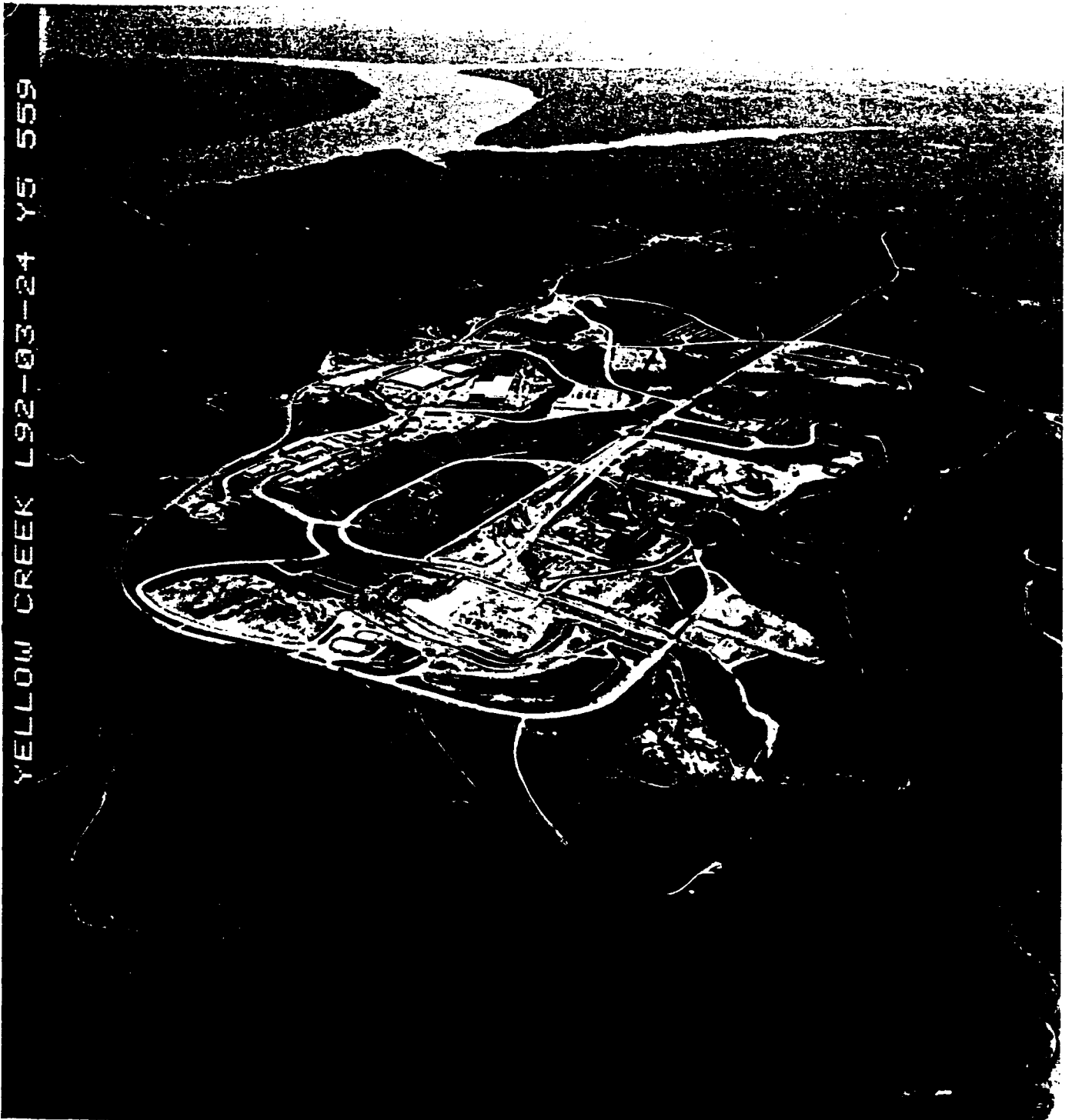
ORIG DATE: 6-JAN-82
 REV B: 15-MAR-82
 STATUS: 13-MAR-82

ASRM SUMMARY PROJECT MASTER SCHEDULE -- OPTION 5A -- FEB 97 LAUNCH

J. W. THOMAS, WB-01

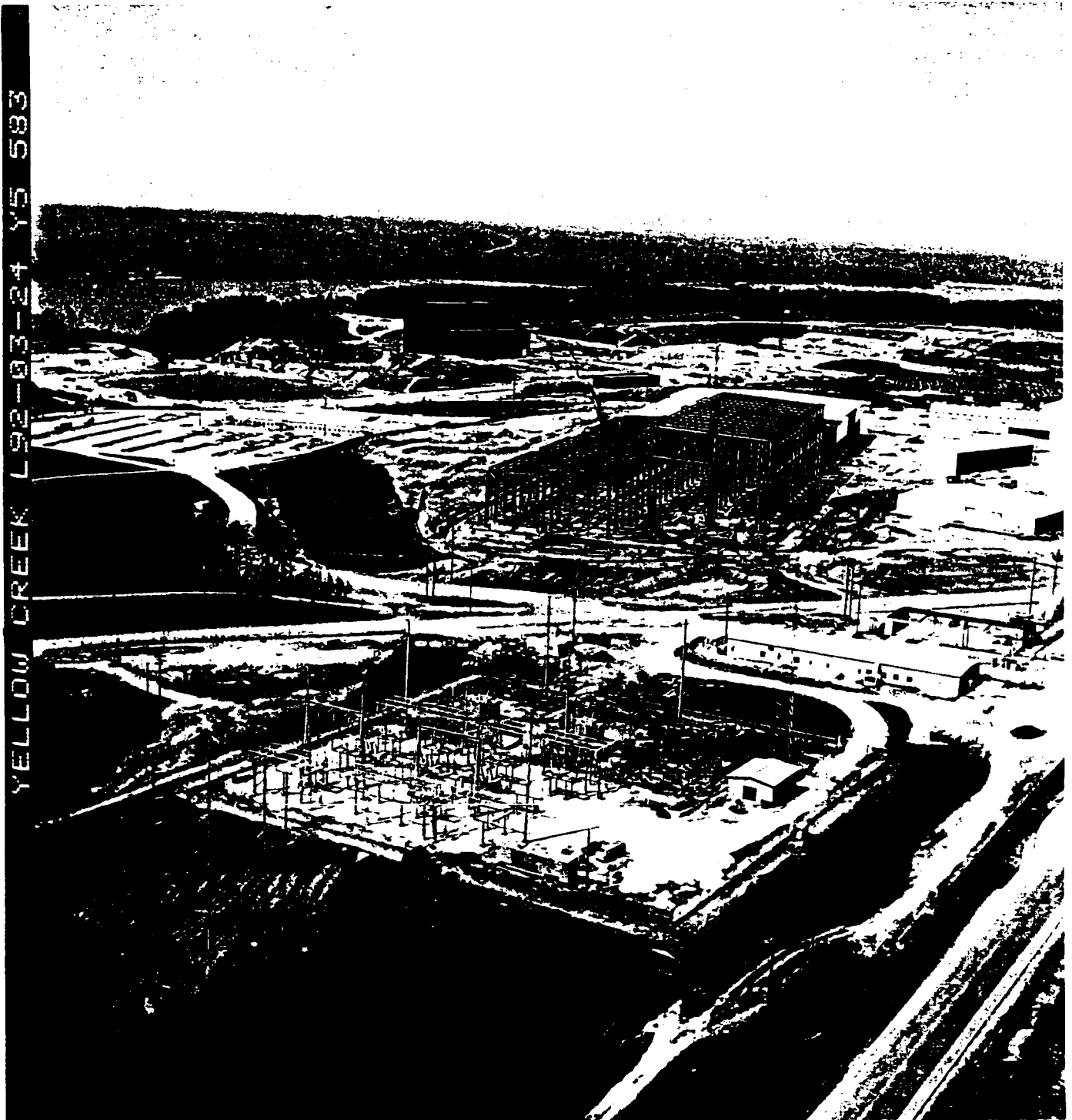


YELLOW CREEK



YELLOW CREEK L92-03-24 Y5 559

CASE PREP BUILDING



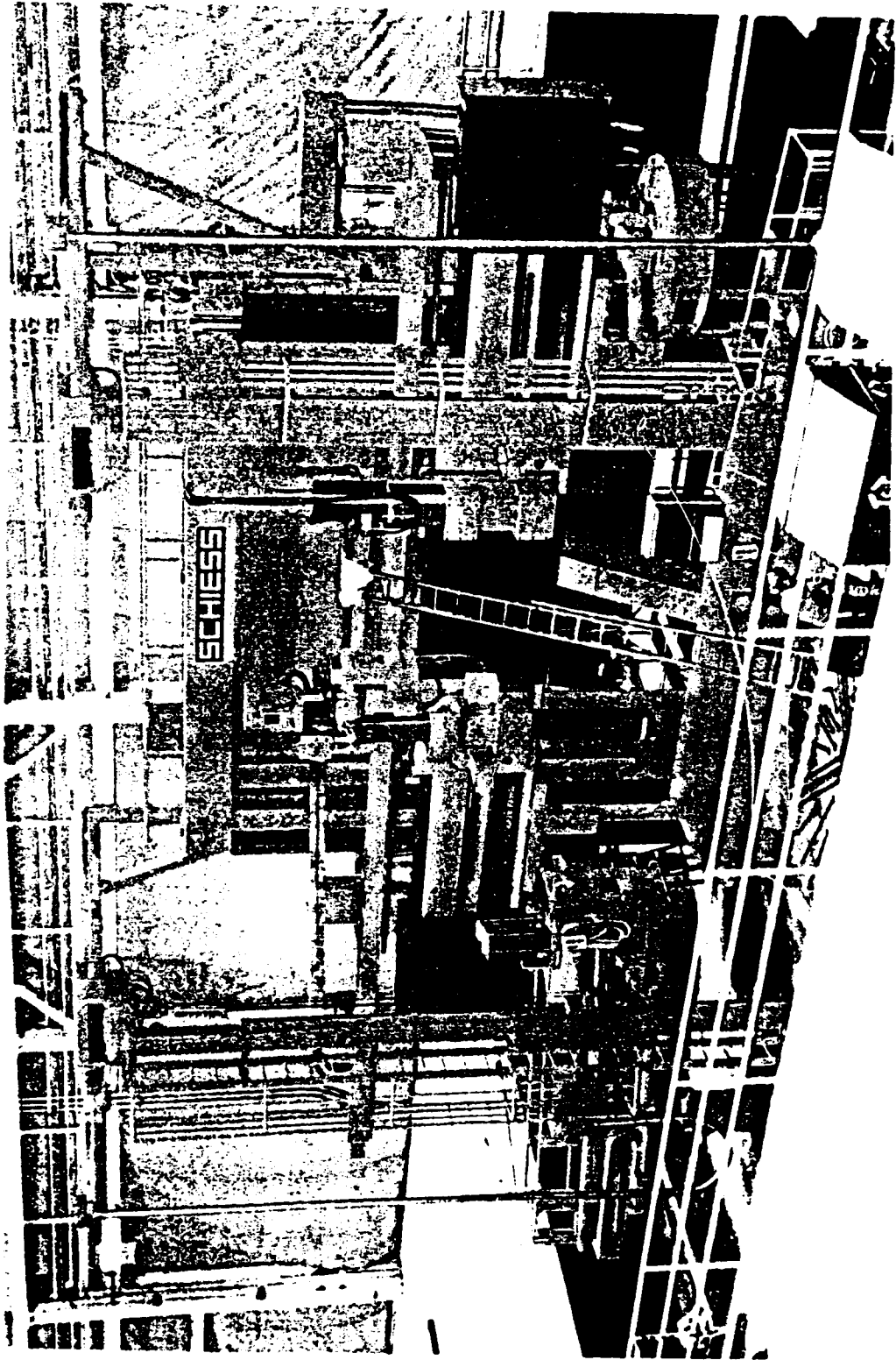
**CASE PREP AUTOCLAVE
ON-SITE FABRICATION AT YELLOW CREEK**



MICHOUD PILING AND PERIMETER WALL

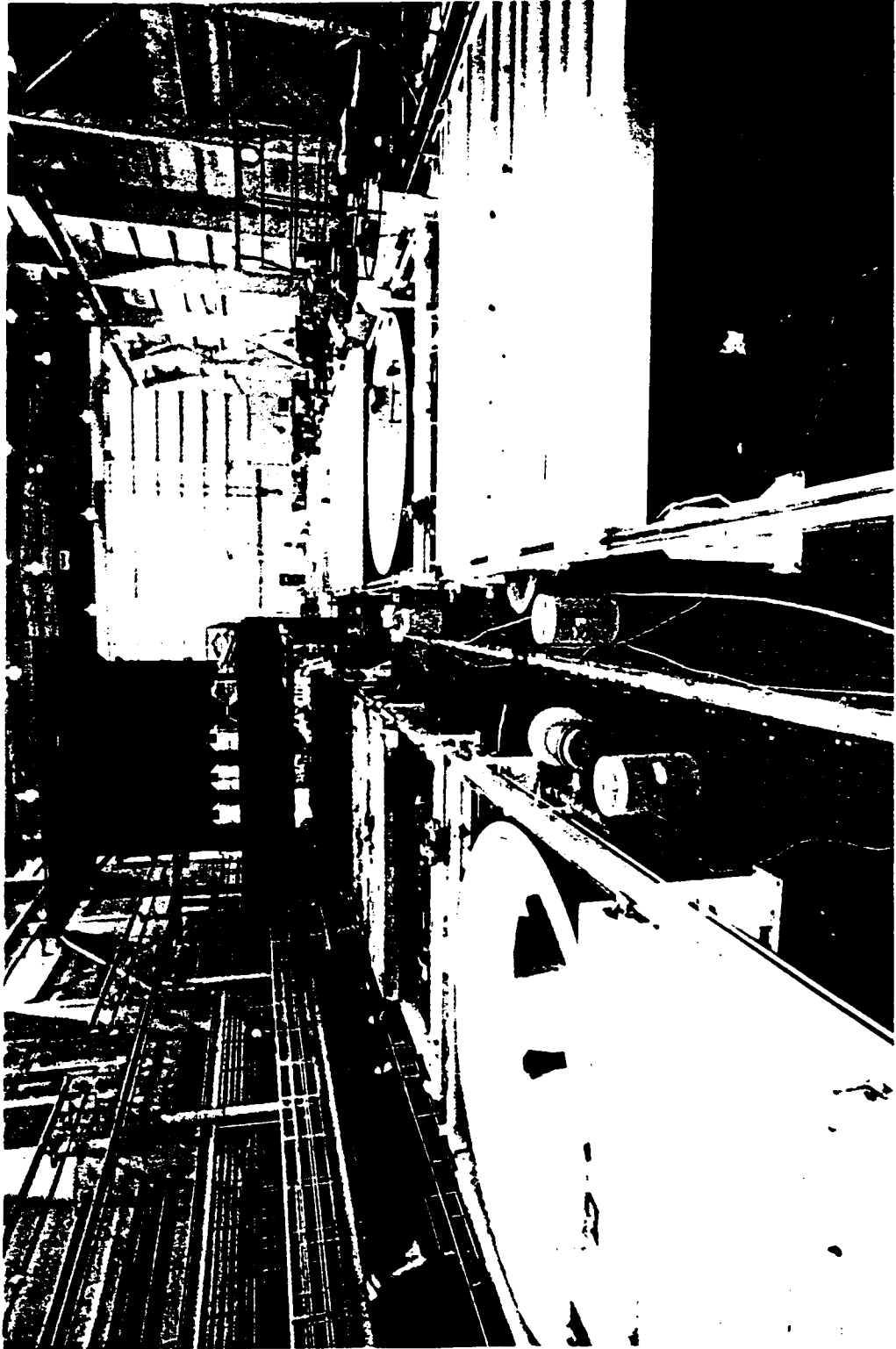


VERTICAL BORING MILL (VBM)



ORIGINAL PAGE IS
OF POOR QUALITY

HEAT TREAT/CHILLER BUILDING



SSC ASRM TEST SITE



ASRM TECHNICAL TOP 5

March 27, 1992

- 1. Mix/Cast Construction, Outfitting & Process Verification**
- 2. Soluble Casting Mandrels**
- 3. Integration Effects; P_c Dot; Loads; Recovery; Overpressure; Moldlines**
- 4. Low-Density Nozzle Ablatives Performance**
- 5. Forward-Facing Cast Inhibitor**

SOLUBLE CORE DESIGN

