

NASA CENTER UPDATE JET PROPULSION LABORATORY

**PRESENTED BY
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JPL

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**1992 NASA AEROSPACE BATTERY WORKSHOP
November 17-19, 1992
U. S. Space and Rocket Center
Huntsville, AL**



AGENDA

- FLIGHT PROJECT SUPPORT ACTIVITIES
 - TOPEX
 - MARS OBSERVER
- RESEARCH / DEVELOPMENT AND ENGINEERING ACTIVITIES
 - NiCd MODEL DEVELOPMENT
 - SECONDARY LITHIUM BATTERY DEVELOPMENT
 - SODIUM - NiCl₂ MODERATE TEMPERATURE BATTERY
 - Li-SOCl₂ BATTERIES FOR CENTAUR LAUNCH VEHICLE
 - DIRECT HYDROCARBON / METHANOL FUEL CELLS



TOPEX MISSION / BATTERY DEFINITION

- PRIME CONTRACTOR - FAIRCHILD / McDONNELL DOUGLAS MPS (BATT)
- BATTERY DESIGN
 - MODULAR POWER SUBSYSTEM (3 x 22 CELL 50 Amp-Hr BATT)
 - CELL DESIGN
 - GATES AEROSPACE - NASA STANDARD
 - 16 POS / 17 NEG
 - PELLON 2505 SEPARATOR
 - NONPASSIVATED POS / TEFLONATED NEG
- BATTERY CYCLE REGIME
 - MEDIUM ALTITUDE ORBIT - VARIABLE OCCULTATIONS AND SOME IFULL SUN PERIODS

_____ BATTERY SYSTEMS GROUP _____



TOPEX STATUS

- LAUNCH AUGUST 10, 1992
- BATTERY OPERATIONAL STRATEGY
 - LIMIT PEAK CHARGE CURRENTS TO 20 AMPS (OFFSET ARRAY)
 - LIMIT OVERCHARGE BY MAINTAINING RECHARGE RATIO (C/D) TO 103% @ 0°C (OPERATE AT LOWER V/T LEVELS)
 - AVOID HIGH CHARGE CURRENTS DURING FULL SUN PERIODS (OPERATE AT LOWER V/T LEVELS)
- CURRENT STATUS - NOMINAL OPERATION



MARS OBSERVER MISSION / BATTERY DEFINITION

- **PRIME CONTRACTOR - GE ASTROSPACE**
- **BATTERY DESIGN**
 - **TWO 17 CELL / 42 Amp-Hr BATTERIES**
 - **CELL DESIGN**
 - **GATES AEROSPACE**
 - **13 POS / 14 NEG**
 - **PELLON 2505 ML**
 - **NONPASSIVATED POS / TEFLONATED NEG**
- **BATTERY CYCLE REGIME**
 - **11 MONTH CRUISE**
 - **~ 120 Min ORBIT 41 Min ECLIPSE (max)**
 - **REQUIRE 9000 CYCLES**

BATTERY SYSTEMS GROUP



MARS OBSERVER STATUS

- LAUNCH - SEPTEMBER 25, 1992
- BATTERY OPERATIONAL STRATEGY
 - DEVELOP METHOD FOR MINIMIZING EFFECT OF 850 mA TRICKLE CHARGE DURING CRUISE
 - MINIMIZE TRICKLE CHARGE BY BATTERY SWITCHING - SWITCH ONE BATTERY OFF LIE FOR 12 HOURS AND THEN REVERSE
- CURRENT STATUS - NOMINAL PERFORMANCE

NiCd MODEL DEVELOPMENT

OBJECTIVE: TO DEVELOP A NiCd BATTERY PERFORMANCE MODEL BASED ON FUNDAMENTAL ELECTROCHEMICAL PRINCIPLES AND CAPABLE OF PREDICTING BATTERY VOLTAGE UNDER SPACECRAFT OPERATING CONDITIONS OVER MISSION LIFE

STATUS: BEGINNING OF LIFE BATTERY LEVEL PREDICTION MODEL IS OPERATIONAL - CELL DESIGN ENGINEERING DATABASE DEVELOPED ALLOWING FOR COMPREHENSIVE CELL SPECIFICATION

PLANS: INCORPORATION AND VERIFICATION OF DEGRADATION FEATURES - FINALIZE DOCUMENTATION - SUBMIT FOR FIELD EVALUATION



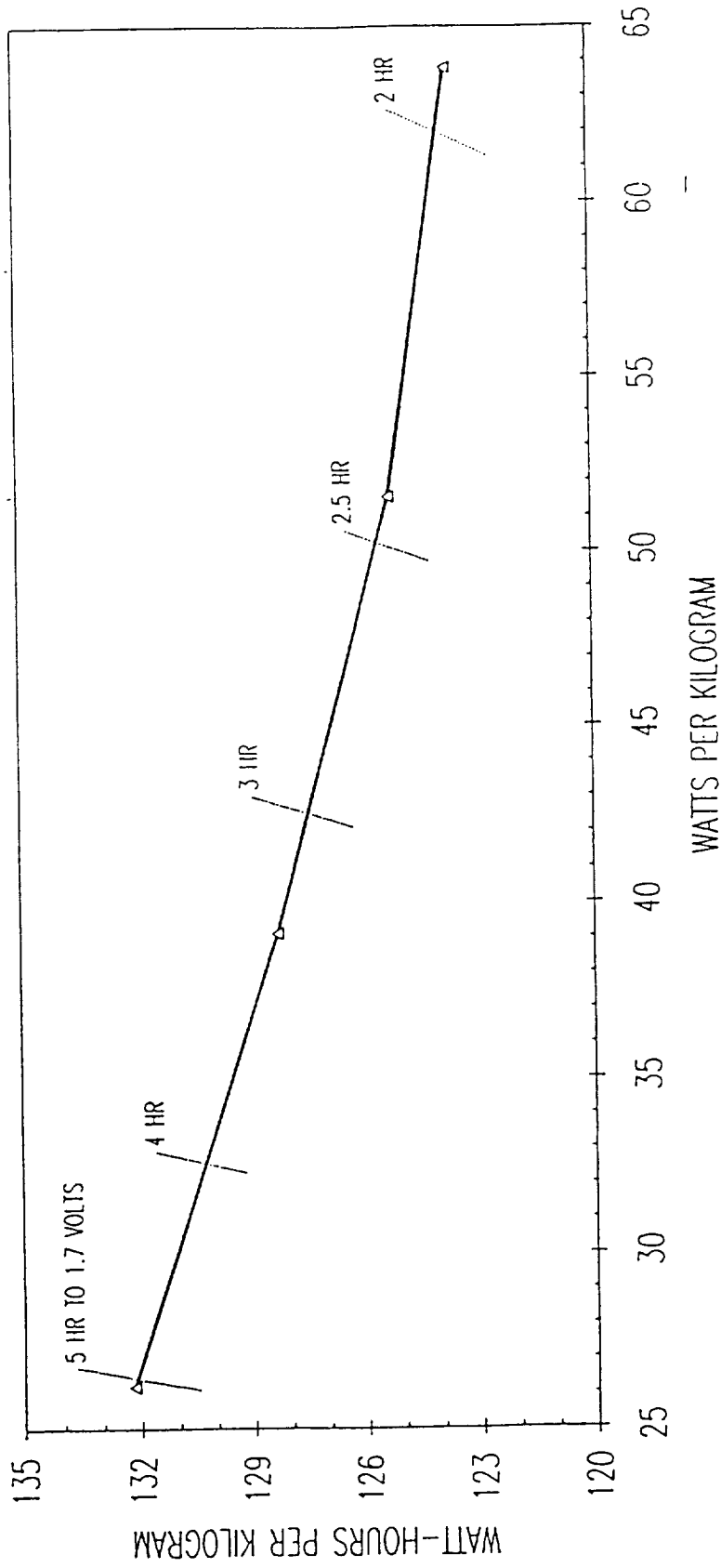
SECONDARY LITHIUM CELLS

OBJECTIVE: TO DEVELOP AND DEMONSTRATE A 100 WH/Kg
LiTiS₂ RECHARGEABLE CELL CAPABLE OF 1000 CYCLES AT 50
% DEPTH OF DISCHARGE AND A 5 YEAR STORAGE LIFE

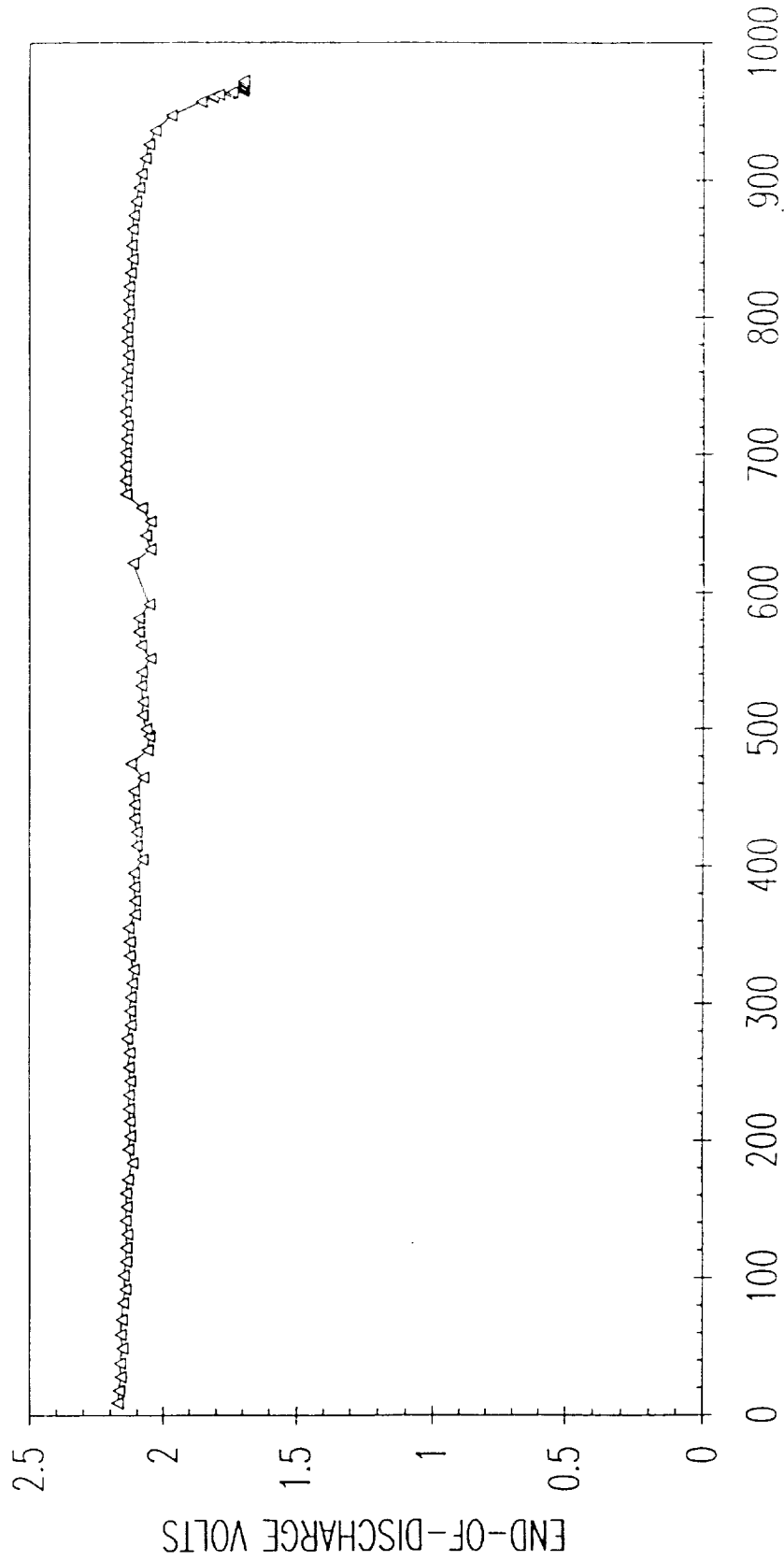
STATUS: 965 CYCLES AT 50 % DEPTH OF DISCHARGE IN 1
AH 'AA' SIZE LiTiS₂ CELLS AT 50% DOD - DEMONSTRATED 125
WH/Kg

PLAN: DEMONSTRATE 1000 CYCLES IN 5 AH PRISMATIC
CELLS - VERIFY OVERCHARGE MECHANISM - COMPLETE
SAFETY TESTING - DETERMINE OPERATING LIMITS

PERFORMANCE OF A TYPICAL 1 AH (AA) LITHIUM TITANIUM DISULFIDE CELL

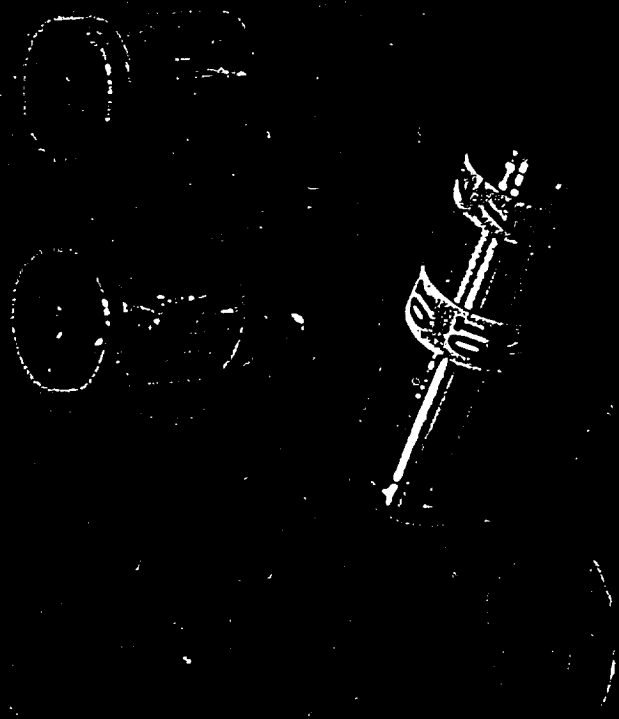


CYCLE LIFE PERFORMANCE OF A 1 AMPERE-HOUR AA LITHIUM-TITANIUM DISULFIDE CELL



50% DOD CYCLES (C/5 DISCHARGE AND C/10 CHARGE TO 2.6 VOLTS)

751-1080



LITHIUM / LIQUID ELECTROLYTE CELL THIS, CATHODE

CHAMBER FOR CATHODE



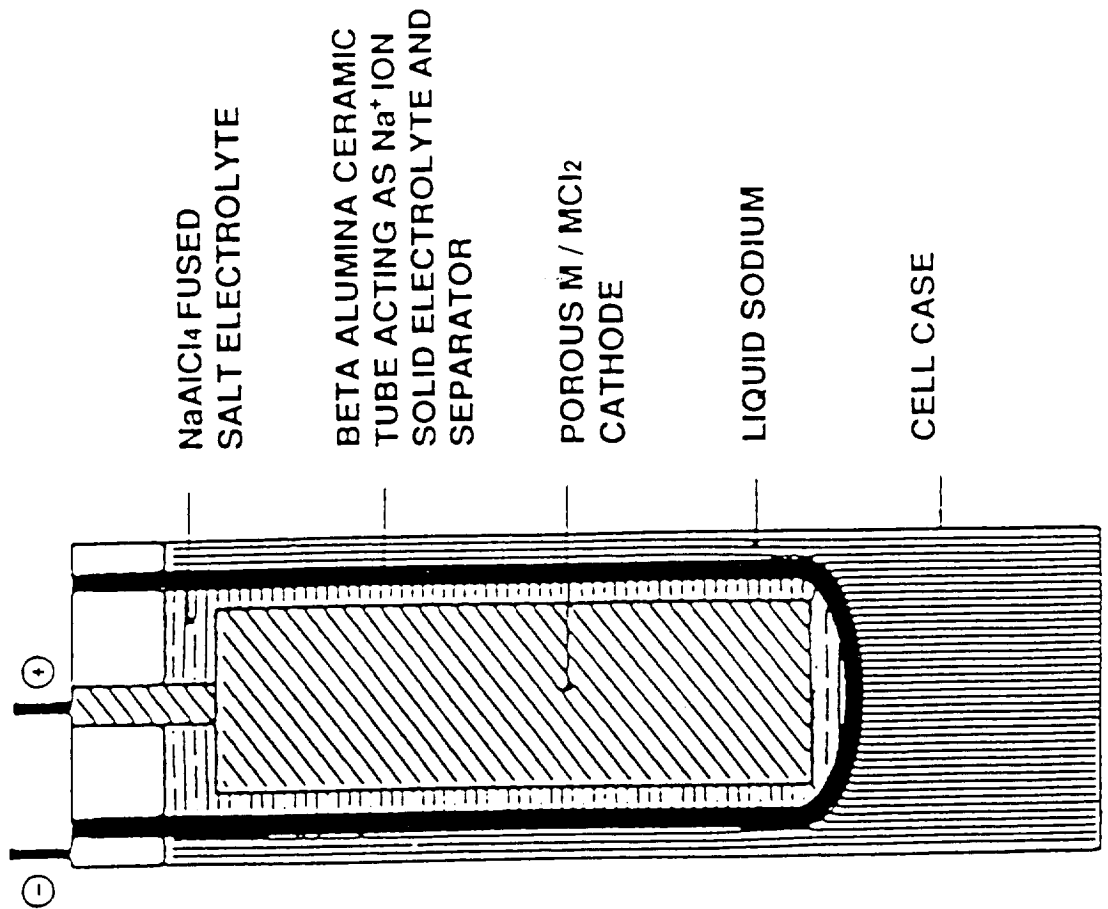
SODIUM METAL HALIDE CELLS

OBJECTIVE: TO DEVELOP A HIGH SPECIFIC ENERGY (>150 WH/Kg) BATTERY FOR FUTURE NASA SPACE MISSIONS

STATUS: NiCl_2 SELECTED FROM SEVERAL METAL CHLORIDES - EXPERIMENTAL CELL TESTS REVEAL OPTIMAL OPERATION AT 275°C - LONG CYCLE LIFE IN LEO CYCLING INDEPENDENTLY VERIFIED AT IN ESA SPONSORED TESTS

PLANS: FABRICATE LABORATORY CELLS AND INVESTIGATE LIFE LIMITING MODES

JPL ADVANCED BATTERY CONCEPTS SODIUM-METAL HALIDE CELLS



CONFIGURATION

Na (l) / BETA ALUMINA (s) / NaAlCl₄ (l) / MCl₂ (s)

CELL REACTION

Na = Na⁺ + e⁻ (ANODE)

MCl_n + n e⁻ = M + n Cl⁻ (CATHODE)

MCl_n + n Na = M + n NaCl (OVERALL)

WHY Na-MCl₂ BATTERIES?

- HIGH ENERGY AND POWER DENSITIES COMPARABLE TO Na - S BATTERIES
- SEVERAL POTENTIAL ADVANTAGES OVER Na-S
- LONG CYCLE AND ACTIVE LIFE
- SEVERAL IMPROVED CATHODE MATERIALS POSSIBLE

| ACTIVITY | 88 89 90 91 92 93 94 95 96 97 98 99 2000 | |
|---|--|---|
| <p>SCREENING STUDIES</p> | <p>Evaluate organic and inorganic cathodes Down select to Na/MCl₂</p> <p>Short term studies performance and reversibility Identify suitable materials Identify and overcome rate limiting processes Down select to Na/NiCl₂</p> | <p>IDENTIFY SYSTEM CAPABLE OF PROVIDING > 1000 CYCLES AND 150 Wh/Kg</p> |
| <p>ELECTROCHEMICAL CHARACTERIZATION OF MCl₂</p> | | <p>ESTABLISH MECHANISMS DETERMINE REACTION KINETICS AND IDENTIFY RATE LIMITING PROCESSES</p> |
| <p>COMPONENT DEVELOPMENT</p> | <p>Develop 5 Ah TEST CELL</p> <p>Study of performing enhancing additives Develop cathode fabrication process Identify cell failure mechanism Charge methods Optimize and improve design</p> <p>Develop performance data base Evaluate safety and environmental effects Identify failure modes</p> <p>Develop eng model cell Demo 1000 cycles and 150 Wh/Kg</p> | <p>DEFINE DESIGN REQUIREMENTS FOR 20-25 Ah CELLS</p> |
| <p>PERFORMANCE AND SAFETY EVALUATION</p> | | <p>DEMONSTRATE CYCLE LIFE AND PERFORMANCE IN OPTIMIZED 20-25 Ah CELL</p> |
| <p>PROTO TYPE</p> | | <p>FINAL DEMONSTRATION</p> |

**250 AH Li - SOCl₂ BATTERY FOR
THE CENTAUR LAUNCH VEHICLE**

OBJECTIVE: TRANSFER JPL DEVELOPED Li-SOCl₂ BATTERY TECHNOLOGY TO 2 CONTRACTORS, FABRICATE CELLS AND BATTERIES AND DEMONSTRATE CAPABILITY FOR MEETING CENTAUR QUALIFICATION REQUIREMENTS

STATUS: DOWN SELECTED TO YARDNEY TECHNICAL PRODUCTS - 5 BATTERIES READY FOR QUALIFICATION - 48 CELLS SUBJECTED TO CHARACTERIZATION TESTS (TEMP, RATE) AND PERFORMED WELL - PDR's. CDR's, AND MRR's COMPLETED

PLANS: COMPLETE CELL / BATTERY TESTS PER CENTAUR REQUIREMENTS - COMPLETE DOCUMENTATION AND DELIVER MCD TO AIR FORCE

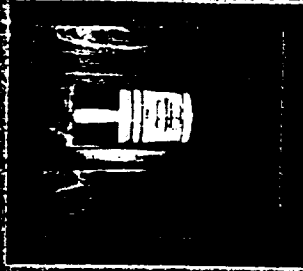
JPL CENTAUR LI-SOCl₂ BATTERY

**ALLIANT
VERSION**



28V 250AH CELL

**YARDNEY
VERSION**



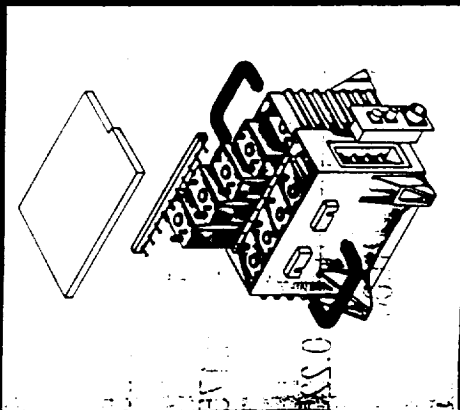
34V 250AH CELL

FEATURES

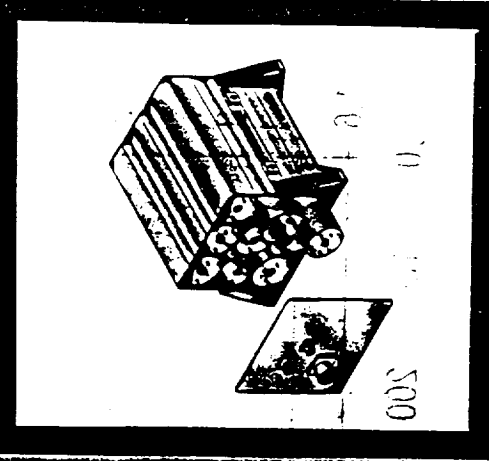
- WEIGHT 76LB (34.5KG)
(1/2 OF EXISTING SILVER-ZINC BATTERY)
- LOW TEMPERATURE LIFE (5YR @ 0°F)
- AMBIENT TEMPERATURE LIFE (1Y @ 40-80°F)
(10 TIMES EXISTING SILVER-ZINC BATTERY)
- CURRENT
 - CONTINUOUS > 40A
 - SHORT TERM > 75A

STATUS

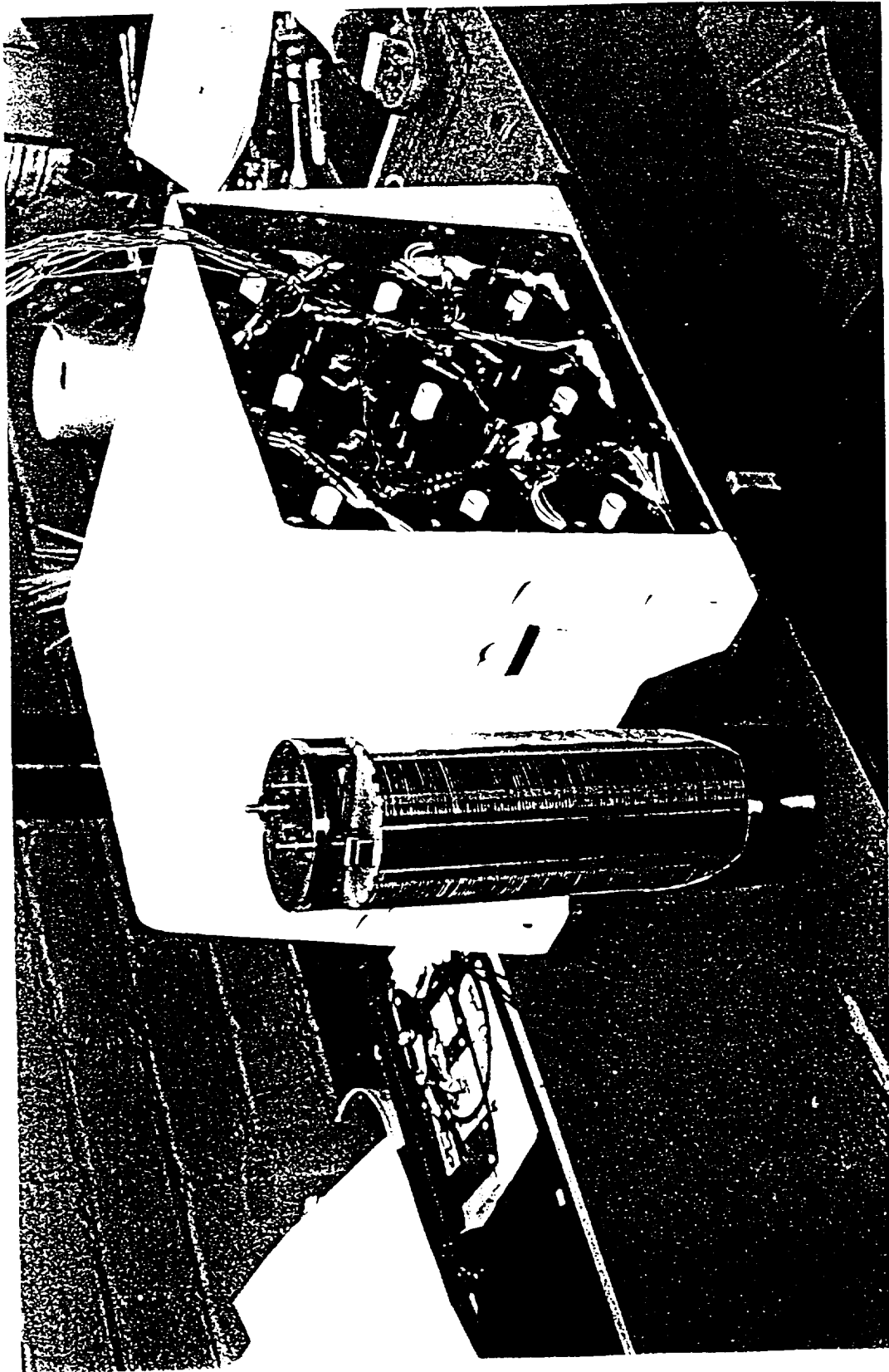
QUALIFICATION
(OF DESIGN
AND MFG) 98%



28V 250AH BATTERY



28V 250AH BATTERY





DIRECT HYDROCARBON / METHANOL FUEL CELLS

OBJECTIVE: TO DEVELOP A FUEL CELL SYSTEM CAPABLE OF THE DIRECT OXIDATION OF METHANOL, METHANE OR OTHER HYDROCARBON

STATUS: NEW CONCEPT IN FUEL CELLS (LIQUID FEED FUEL CELL) HAS BEEN DEMONSTRATED AND ACHIEVED 80 mA/cm² AT 0.5 VOLTS

PLANS: CONTINUE THE DEVELOPMENT AND EVALUATION OF LIQUID FEED FUEL CELL - EVALUATE NEW CATALYSTS - FAB DEMONSTRATION UNIT

HIGHLIGHTS OF THE JPL DARPA DIRECT METHANOL FUEL CELL TASK

OBJECTIVE

DEVELOP DIRECT METHANOL FUEL CELL TECHNOLOGY (DMFC) AT THE CELL LEVEL WITH TARGET PERFORMANCE LEVELS BY 1994

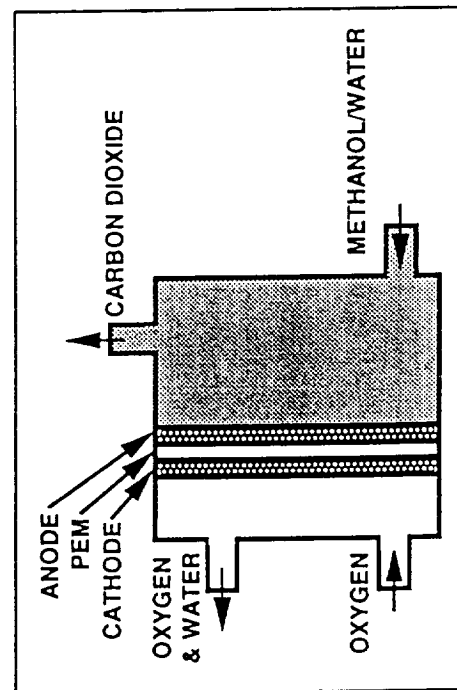
TARGETS (CELL LEVEL)

CURRENT DENSITY > 150 mA/cm²
 CELL VOLTAGES > 0.6 VOLTS
 LIFE > 1000 HOURS
 TEMPERATURE < 200°C

ACCOMPLISHMENTS

- SELECTED CATALYSTS AND ELECTROLYTES WITH INPUTS FROM UNIVERSITIES FOR THE INTERIM METHANOL/O₂ SYSTEM DEMONSTRATION
 - PT/RU
 - NAFION MEMBRANE
 - C8 ACID
- IDENTIFIED LIQUID FEED DESIGN AS ATTRACTIVE FOR LOW TO MEDIUM POWER APPLICATIONS
- DEMONSTRATED FEASIBILITY OF LIQUID FEED DESIGN WITH SUPPORT FROM GINER 0.54V AT 100 mA/cm²
- EVALUATION OF ALTERNATIVE FUELS IN PROGRESS
 - TRIMETHOXYMETHANE
 - DIMETHOXYMETHANE

SCHEMATIC OF LIQUID FEED CELL



ADVANCES IN DIRECT METHANOL FUEL CELLS

