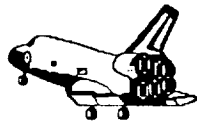


V93-22108

-REENTRY SYSTEMS- MATERIAL TECHNOLOGY NEEDS



R. M. (MIKE) EHRET
M&P ENGINEERING & LABS
SPACE SYSTEMS DIVISION
9/24/91



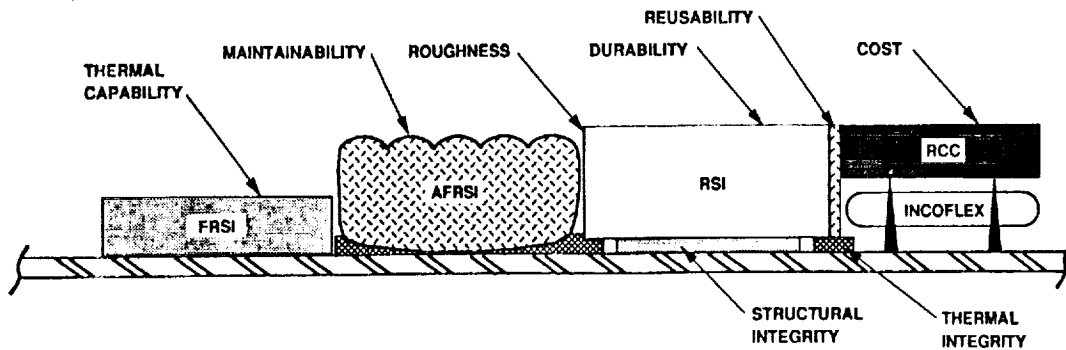
Rockwell International
Space Systems Division

BACKGROUND IN ENTRY SYSTEMS

- MIKE EHRET - MATERIALS ENGINEER
- 23 YEARS ROCKWELL SPACE DIVISION
 - SATURN S-II
 - SPACE SHUTTLE ORBITER
- MANAGER: MATERIALS & PROCESSES
 - ENGINEERING & LABORATORIES
- ENTRY SYSTEMS BACKGROUND
 - STRAIN ISOLATION
 - TILE DENSIFICATION
 - FRCI TILE CERTIFICATION
 - AFRSI DEVELOPMENT
 - WATER PROOFING
- PERSONAL PERSPECTIVES:
 - DESIGN (PERFORMANCE)
 - BUILD
 - OPERATIONS
 - MAINTAINABILITY

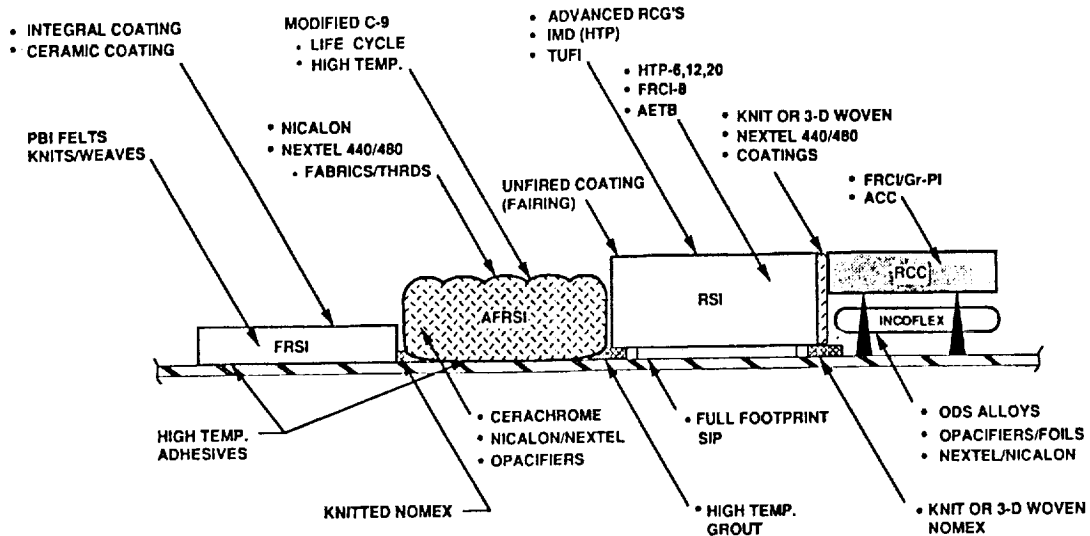
POTENTIAL IMPROVEMENTS EXIST WITHIN CURRENT ORBITER TPS SYSTEM

700 F	1,500 F	2,300 F	3,200 F
\$650/FT ²	\$2,000/FT ²	\$10,000/FT ²	\$30,000/FT ²
0.15 - 0.25 LB/FT ²	0.62 - 1.25 LB/FT ²	0.90 - 3.5 LB/FT ²	7.4 LB/FT ²
3,000 FT ²	3,000 FT ²	5,000 FT ²	400 FT ²



EXISTING SYSTEM IS FUNCTIONAL BUT MAY NOT BE MOST COST-EFFECTIVE

ADVANCED TPS OPPORTUNITIES



TPS MATERIAL ENHANCEMENTS ARE FEASIBLE

MATERIAL/CONCEPT	BENEFITS	TECHNOLOGY GAPS	TRENDS
RIGID TPS: (i.e., AETB, HTP, ACC-HARDSHELL, METALLIC STANDOFF, TUF1 COATING, TITANIUM MULTIWALL, IMD, SOL-GEL RCG)	<ul style="list-style-type: none"> HIGHER STRENGTH HIGHER TEMPERATURE IMPACT RESISTANT LIGHTER WEIGHT ADJUSTABLE DENSITY 	<ul style="list-style-type: none"> PRODUCTION SCALE-UP AVAILABILITY MAINTAINABILITY COATINGS COATINGS APPLICATION INDUSTRY DATA BASE MECHANICAL PROPERTIES INSTALLATION PROCEDURES 	<ul style="list-style-type: none"> LIGHTER WEIGHT DURABLE COATINGS MATERIAL CONSISTENCY HIGHER TEMPERATURE TAILORED DENSITIES STRONGER
FLEXIBLE TPS: (i.e., TABI, PBI)	<ul style="list-style-type: none"> INCREASED TEMPERATURE TAILORABLE PROPERTIES PRODUCT FORMS LOWER COST THAN RIGID REDUCED VULNERABILITY 	<ul style="list-style-type: none"> PRODUCTION SCALE-UP COATINGS IN-SERVICE USE INDUSTRY DATA BASE 	<ul style="list-style-type: none"> CONSTRUCTION METHODS FIBER TREATMENT OPTIMIZATION MIXING FIBER BLENDS USED IN LIEU OF RIGID HIGHER TEMPERATURE
FOAMS/ABLATORS: (i.e., SOFI, NCFI, SLA 561, POLYIMIDE, POLYMETHACRYLIMIDE)	<ul style="list-style-type: none"> LOWER COST vs TILE FORMABLE HIGH DIMENSIONAL STABILITY UNDER HEAT FIRE RESISTANCE EXCELLENT RADIATION TRANSMISSION 	<ul style="list-style-type: none"> IMPROVED MECHANICAL PROPERTIES AT ELEVATED TEMPERATURE LIGHTWEIGHT SANDWICH CONSTRUCTION PRODUCTION SCALE-UP AVAILABILITY INDUSTRY DATA BASE 	<ul style="list-style-type: none"> NON-CFC BLOWN LIGHTER WEIGHT IMPROVED HEAT TRANSFER PROPERTIES IMPROVED FABRICATION
REFRACTORY COMPOSITES: (i.e., ACC, C-C, SIC, SIC-SIC)	<ul style="list-style-type: none"> HIGH TEMPERATURE LOAD CARRYING AT HIGH TEMPERATURE WEIGHT SAVINGS DIMENSIONALLY STABLE 	<ul style="list-style-type: none"> INSPECTION COATING REPAIR HIGH TEMP COATINGS LOW COST JOINING COMPLEX STRUCTURES IN-SERVICE 	<ul style="list-style-type: none"> OXIDATION RESISTANCE THERMALLY STABLE FIBERS IMPROVED MATRIX AUTOMATED PROCESSING

SUMMARY OF TECHNOLOGY NEEDS AND DIRECTION

NEEDS

- **LIGHTWEIGHT AND DURABLE RIGID INSULATION AND HIGHER TEMPERATURE FLEXIBLE MATERIALS**
- **INSPECTION, REPAIR, PRODUCIBILITY, AND MAINTAINABILITY OF REFRACTORY COMPOSITES**

DIRECTION OF EFFORTS

- **FUNDING BASE IS RELATIVELY SMALL FOR FUTURE YEARS**
- **TO MAXIMIZE RETURNS, COLLABORATIVE PROGRAMS APPEAR TO BE PRACTICAL**
 - **SSD'S APPROACH IS TO IMPLEMENT NASA DEVELOPED TECHNOLOGY**

SPACE TRANSPORTATION STRUCTURES AND MATERIALS WORKSHOP

ENTRY SYSTEMS PANEL

- **DON'T DESIGN A SPACECRAFT AS THOUGH IT WILL BE TREATED LIKE A SPACECRAFT**
- **DON'T BELIEVE PRELIMINARY LOADS**
- **DON'T ALLOW MATERIALS R&T HISTORY TO VANISH**
- **DON'T CERTIFY WITHOUT SYSTEM LEVEL TESTS**
- **DON'T BELIEVE THAT THE DESTROYER OF "GOOD" IS "BETTER"**
- **DON'T BUILD ANYTHING NEW WITH SOA MATERIALS TECHNOLOGY**