AIRCRAFT CIRCULARS

NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS

No. 91

THE DE HAVILLAND D.H. 75 "HAWK MOTH" (BRITISH)
Cabin Monoplane

Washington March, 1929 NATIONAL ADVISORY COMMITTEE FOR AERONAUTICS.

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Cabin Monoplane

The De Havilland "Hawk Moth" represents an attempt to give in the air the comfort which one usually associates with a good motor car. It is a four-seater, with the occupants placed two by two. And in the cabin arrangement, one traces one of those "outside" considerations which may outweigh the merely aircraft ones. From the aerodynamic point of view it would have been preferable to make a very narrow fuselage, and quite conceivably a good many horsepower would have been saved by such a body. But, unfortunately, an airplane has to be something more than merely an aircraft. has to carry people in reasonable comfort. In the "Hawk Moth" the seating accommodation is comfortable. The cabin is wide enough for two to sit side by side without crowding. And there is room to stretch one's legs, as well as to keep them in a number of different positions. Nothing is more tiring than to sit for two or three hours in a narrow seat in which there is no elbowroom, and with one's legs in one particular position only.

The sloping wind screen in front gives a very good view forward. The large side windows and the absence of a lower wing affords the occupants an unobstructed view of the ground.

^{*}From Flight, February 7, 1929.

and the roof light, which forms the top of the cabin, enables the pilot to look back and up to ascertain if he is being overtaken by another airplane. What is the aerodynamic effect of not continuing the wing across the fuselage we do not know. It may be small and it may be large, it may be favorable or unfavorable. But it does give a very excellent view upwards, and makes the cabin quite unusually light and cheerful. The mental effect of this is likely to be considerable. subjects as ease of handling, performance, etc., it is too early to express an Epinion yet. The airplane has but recently been finished, and officially checked performance figures are not yet available. The preliminary test flights have indicated that the "Hawk Moth" handles very well, and that it has appraximately the performance which had been expected. instance, the cruising speed is likely to be in the neighborhood of 100 M.P.H., which is a very useful speed, and suffie' cient for most purposes at present.

Constructional Details

Structurally, the "Hawk Moth" is a composite, with a metal fuselage and wooden wings, although it is to be presumed that a metal wing is a likely future development (Fig. 1). In the construction of the fuselage use is made of steel tubing of square and circular section. All longerons are of square section, but in the rear portion the struts are of round section,

welded to the longerons, while in the front square section tube is used for the struts also, and bolted joints are used instead of welded. The system of construction is based upon the manufacture of the sides as complete units, these being built up "flat," and blocks have their inner ends drilled conically, the effect being to leave four tapering prongs in the corners of the tube.

The tail surfaces are of steel tube construction, welding being used to a considerable extent for jointing, as shown in Figure 4. The rudder does not, as is common practice, extend down to the bottom end of the sternpost. Instead, the fuse-lage terminates in a form of "cruiser stern," with the rudder wholly above it, and the extension of the fuselage houses the and 3 tail skid. In Figures 1/a tail wheel is shown, but actually this will be supplanted by a tail skid carrying a wheel at its end. This change has been decided upon as a result of the possibility of the stern being damaged owing to the steep angle of the supporting pillar and the small diameter of the wheel, which, on meeting a hard obstacle, might receive a knock in a horizontal direction.

The wings, as already indicated, are of wood construction, with box spars and orthodox wooden ribs, fabric covered. The two halves of the wing are hinged to the top corners of the fuselage, the front spar joint having a quick-release pin and the rear spar a hinge, as the wings are designed to fold.

The wing is braced by two sloping struts on each side, attached to fittings on the lower longerons. A fairly short diagonal strut runs from front to rear main strut, and serves to stabilize the wing structure when the wing is folded. A telescopic jury strut is permanently hinged at one end, with the other held in a catch when the strut is not in use. This jury strut serves, when the wing is folded, to support the forward corner.

The landing gear is of very wide track, and rubber blocks working in compression form the shock-absorbing medium. The telescopic leg is supported at the top by Vee tubes to the fuselage, and at the lower end another Vee is formed by the bent axle and the radius rod, as shown in Figure 5.

The power plant of the first airplane is one of the new de Havilland "Ghost" engines, a Vee type air-cooled of eight cylinders (Fig. 2). In effect this engine is two "Gipsies" placed together in Vee formation. Reduction gearing is employed, and the engine delivers some 200 HP. For those who desire to use a radial engine, the Armstrong-Siddeley "Lynx" can be fitted, a suitable "nose" to take this power plant having been designed.

The gasoline tanks are placed inside the wing, one on each side, and give direct gravity feed. The fuel capacity will be about 35 gallons in each tank, which would give the airplane a duration of something like eight hours. Normally a smaller quantity would probably be carried.

The pilot occupies the forward seat on the port side, and the controls are of somewhat unusual type in that the lateral movement of the "stick" pivots about a point some distance up; the lower portion, with its sprockets and chain, being enclosed in a casing (Fig. 6). Pedals are used instead of the more usual foot bar, and not only is provision made for adjusting the pedals, but the cross par that supports them may be locked in any position by a simple friction device (Fig. 4). Wheel brakes are fitted, and can be operated either together or independently, thus facilitating maneuvering on the ground. The cabin is heated by a muff around the exhaust pipe, and ventilation is provided. The side windows are of the sliding type, and can be opened to a greater or smaller degree as desired.

Characteristics

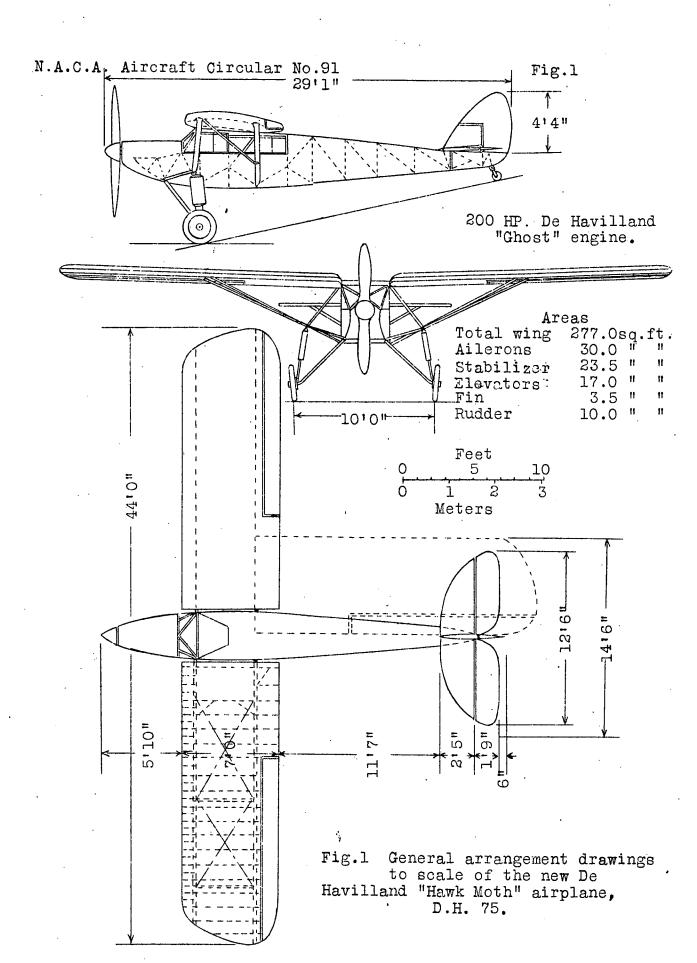
Length

29 ft. 1 in.

Span	44 "	. O ii
Areas:		
Total wing	277.0	sq.ft.
Ailerons	30.0	tt
Stabilize r	23.5	!! .
Elevators	17.0	11
Fin	3.5	!!
Rudder	10.0	11

Characteristics (Cont'd)

Weight (tare)	about	2000	lb.	910	kg
Total loaded weight		3500	Ħ	159 0	11
Gross weight - normal - (made up of 4 people, 200 lb. of luggage, and about 35 gal. of gasoline and oil)	about	3200	II	1455	tī
Wing loading (for normal gross weight)		11.	. 56	lb./sq.ft	•
Power loading		. 16.	.00	lb./HP.	
Fuel consumption (normal estimated consumption at a cruising speed of about 100 M.P.H., which represe a mileage of approximated 12 miles per hour).	ents	8	to	8.5 gal./	nr.



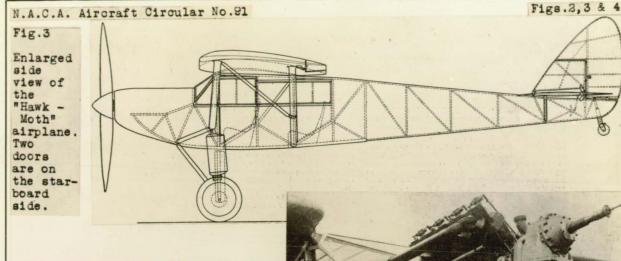


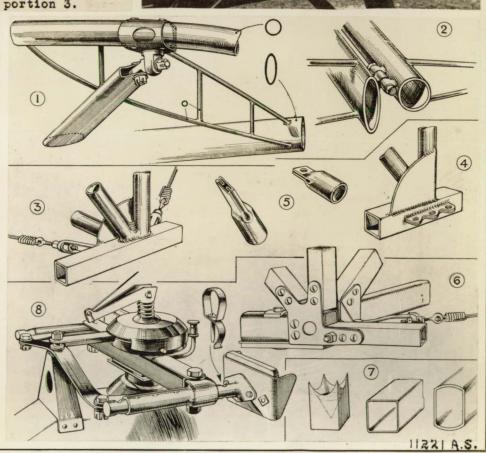
Fig.2 The latest De Havilland engine the "Ghost" as fitted in the "Hawk Moth" airplane.

This engine is of the 8 cylinder V type and is air-cooled.

Fig. 4 Welded tubular construction is employed in the "Hawk - Moth" tail structure 1. Elevator 2 and fuselage rear portion 3.

The latter, it should be noted, uses square-section longerons and round section struts.

The side panels are made up as units, and the top and bottom struts complete the structure 4 & 5. In the front portion square-section struts are used and bolted joints 6, the strut ends being reinforced by aluminium packing blocks shaped to reduce the section gradually by conical drilling 7. To relieve the pilot on long flights, the rudder bar can be locked in position, as shown in 8.

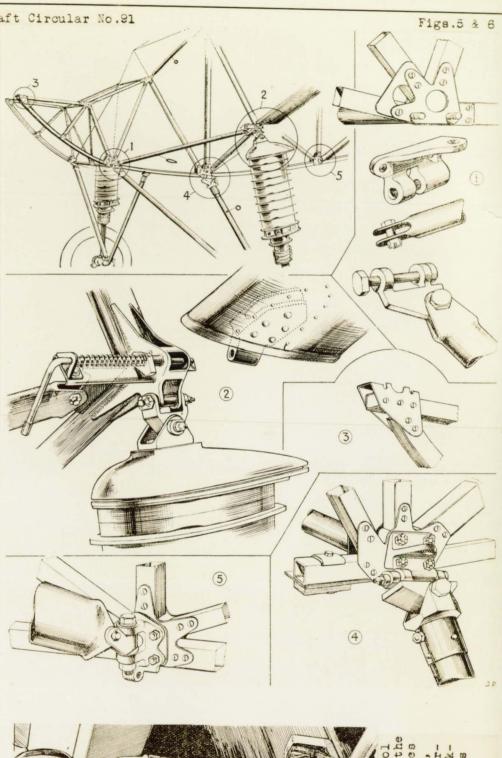


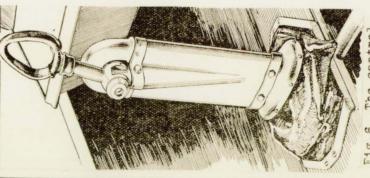
N.A.C.A. Aircraft Circular No. 21 Fig.5 Some constructional details of the "Hawk -Moth". The diagram in the upper left corner indicates the location of the various details illustrated. The joint of the lower longeron to landing gear and wing struts, etc., is shown in 1, the use of steel -

leg and front wing strut. For folding the wing the L shaped spring-loaded pin is withdrawn, and the wing strut is swung with its lower end over to the back strut, where a fitting receives it. The meeting of lower longeron and engine mounting is illustrated in 3, while 4 shows attachment of bent axle and diagonal strut to lower longeron. Finally the joint between rear wing strut and lower longeron

is shown in 5

tube longerons of square section should be noted. 2 shows the attachment of landing gear





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