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B-70 AIRCRAFT

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This document is submitted in compliance with NAS9-12100

SD 72-SH-0003 B-70 AIRCRAFT STUDY FINAL REPORT Volume 11 April 1972 K.J. Taube Study Manager

Study Manager B-70 Aircraft Study



## VOL:II

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J. TaubeStudy ManagerB-70 Aircraft Study





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#### AIR VEHICLE WBS 1.0

#### GENERAL

The B-70 Program, at the onset, was a full weapon system capable of sustained Mach 3 flight for the major portion of its design missions. The weapon system was to enter the SAC inventory as an RS-70 with the first intercontinental reconnaissance/bomber wing scheduled to go operational in July, 1964. After several redirections, a two XB-70 air vehicle program emerged with its prime objective being to demonstrate the technical feasibility of sustained Mach 3 flight.

This section describes the original Weapon System 110A concepts, the evolution of the RS-70 design, and the XB-70 air vehicles which demonstrated the design, fabrication, and technical feasibility of long range Mach 3 flights at high altitude. The data presented shows that a very large step forward in the state-of-the-art of manned aircraft design was achieved during the B-70 development program and that advances were made and incorporated in every area, including design, materials application, and manufacturing techniques.

#### WEAPON SYSTEM 110A REQUIREMENTS

The operational mission of the full weapon system was the effective accomplishment of the USAF strategic bombardment high-altitude supersonic mission. Particular emphasis was on those portions to which the capabilities of the contemporary strategic missiles were not suited and for which the manned bomber had the only proven capabilities. The man-machine combination was to provide discretionary capabilities for target discrimination, malfunction correction or override, timely evasive maneuvers, judgment in selection and employment of penetration aids, heavy payload of mixed weapons, intelligence collection, damage assessment, best altitude and penetration routes, recallability and recoverability.

The design of the weapon system was to ensure ease of production, easy and rapid installation of subsystems and equipment, and ease of general maintenance with a minimum of elapsed time allowed for maintenance periods. Components and materials were to be selected to reduce the requirements for manpower, skill level, special tools, and ground equipment. The degree of accessibility of the equipment as installed in the air vehicle element was commensurate with the frequency of inspection, servicing, or repair of the item. The offensive subsystem (supported by the defensive subsystem to increase survival probability) was to provide accurate navigation to distant target areas and launch weapons with precision for total target destruction.

In conjunction with the above design requirements, three basic range missions were established as baseline design criteria for the weapon system. These basic range missions are schematically presented on the global picture of



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Exhibit 1, page II-11. As shown, two of the missions penetrated the target area at high altitude with total ranges of 6447 nautical miles unrefueled and 7748 nautical miles refueled. The third mission, which was also inflight refueled, had a total range of 5312 nautical miles with an "on-the-deck" target area penetration at Mach 0.95 for 856 nautical miles.

#### **B-70 DEVELOPMENT**

In 1955 when the Weapon System 110A design studies were initiated and based on the existing state-of-the-art in aerodynamics, the air vehicle was visualized as being very large with tremendous fuel capacity. As presented by Exhibit 2, page II-12, this design included two floating wing tip auxiliary fuel tanks, each the size of a B-47 medium bomber which had to be jettisoned prior to supersonic flight. At the onset, it was estimated that the vehicle would gross at 650,000 lbs.; however, by mid 1956, the gross weight estimate had increased to 700,000 lbs. Due to the logistics of the expendable wing tip tanks (two for each mission) and the need for additional runway construction (each SAC base), it was evident that this configuration was unacceptable, whereby an extensive aerodynamic and high temperature development program was initiated.

The aerodynamic and high temperature development programs resulted in significant advancements in the state-of-the-art for the basic airframe and subsystems. A summary of the aerodynamic features and arrangements developed is presented by Exhibit 3, page II-13, while Exhibit 4, page II-14, summarizes the advancements made in high temperature development. The B-70 air vehicle, as depicted by Exhibit 3, page II-13, had an estimated gross weight of 550,000 lbs. which did not impact the existing SAC bases or present any unacceptable logistic problems. This arrangement or configuration became "baseline" for the RS-70 and subsequently the XB-70 air vehicles. The configuration is discussed in detail under each applicable subsystem as identified by the WBS.

#### RS-70 AND XB-70 CONFIGURATIONS

The RS-70 general arrangement is presented by Exhibit 5, page II-15, which evolved into that presented by Exhibit 6, page II-16, for the XB-70 air vehicles. As shown for the RS-70, the nose section contained the radar bomb-sighting antenna, inertial navigation platform, Doppler radar, and an air refueling receptacle. The cabin contained provisions for a crew of four in encapsulated seats and were identified as pilot, copilot, bombardier-navigator, and defense system operator. The electronic equipment compartment aft of the cabin contained the majority of electronic equipment in racks on either side of an aisleway. This compartment, which was pressurized and air conditioned, was accessible in flight. The environmental conditioning equipment was located aft of the electronic equipment bay immediately forward of the integral fuselage fuel tankage. The central portion of the fuselage contained engine air ducts, multipleweapon bay, landing gear, and integral fuel tankage. The wings, including



the folding tips, provided additional integral fuel tankage and contained six elevons on each side of which the two outboard elevons locked "intrail" when the tips folded. The aft fuselage supported twin verticals, a cluster of three drag chutes, and six General Electric J-93 engines.

The XB-70 air vehicles, as depicted by Exhibit 6, page II-16, retained the essential configuration, fundamental areas, and basic flight equipment of the RS-70. The XB-70 equipment provisions were different from the production versions in the following important aspects.

- 1. Radar bomb-sighting antenna, inertial platform, Doppler radar, and air refueling receptacle were omitted from the nose section.
- 2. The crew consisted of pilot and copilot.
- 3. The electronic equipment retained was installed in the cabin instead of the electronics equipment bay.
- 4. Weapon installation provisions were removed from the bomb bay.
- 5. Wing tip fuel tanks and bomb bay fuel provisions were eliminated.
- 6. The bombing/navigation/missile guidance (BN & MG) and defensive systems were eliminated.
- 7. The mission and traffic control system (M&TC) was replaced by available GFP communication, navigation, and identification equipment.
- 8. Ballast was incorporated in appropriate quantity and location to maintain air vehicle balance and aeroelastic properties that represented the production air vehicle.

#### FLIGHT TEST PROGRAMS

The RS-70 Flight Test Program was planned to be conducted in four phases identified as Category I, II, III and IV. Category I flight test program was to be the prime responsibility of the contractor and included test beds and all prime vehicle development up to and including the full weapon system demonstration. Category II was a 20-flight hour phase scheduled for air vehicle No. 1 early in the flight test program and was to be conducted by Air Force pilots to establish feasibility of the basic air vehicle. Category III was to be conducted by the Air Force using Air Force airborne crews but with maintenance and on-the-job training provided by the contractor. This phase of testing was to be on the full weapon system conducted by the Air Force Test Center to evaluate operational suitability. Category IV was to be a complete Air Force program where operational type air and ground crews evaluated the full weapon system under simulated operational conditions.



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Exhibits 7, 8, 9, and 10, pages II-17, II-18, II-19, and II-20, present a detail breakdown of the planned Category I flight test hours scheduled for the RS-70. Exhibit 7, page II-17, shows the chargeable flight hours to each major task that was to be conducted on the prime air vehicles while Exhibit 8, page II-18, shows the scheduled test bed flight hours. Exhibit 9, page II-19, presents the total chargeable flight hours scheduled for the basic air vehicle development. The RS-70 flight test program was based on the test or time sharing concept which is the most timely and economical means of providing the maximum test experience. This concept requires a large capacity instrumentation system and detail planning starting with wind tunnel tests, ground tests and each air vehicle flight. As shown by Exhibit 10, page II-20, utilizing time sharing, test experience was increased for the basic vehicle from 1575 charge hours to 4280 equivalent test hours. This same percentage increase due to time sharing was used in planning the XB-70 flight test program and was essentially verified during the test program.

The detail breakdown of the XB-70 flight test program is presented by Exhibits 11, page II-21, and 12, page II-22. Exhibit 11, page II-21, shows the charged flight hours for the major subsystems, while Exhibit 12, page II-22, presents the equivalent flight test hours based on the percentage gain due to time sharing. The charged flight hours of Exhibit 11, page II-21, were those planned for the XB-70 program which were within a total tolerance band of less than 6.5 hours when compared to the actual charged hours. Based on this small error and to facilitate analysis, the planned charged hours were used as the basic flight hours charged during the XB-70 program as depicted by Exhibit 12, page II-22. Exhibit 13, page II-23, presents a percentage comparison of the XB-70 program flight hour exposure to that planned for the RS-70 program. The percentages shown are based strictly on flight hours exposure and do not reflect the fidelity of the subsystem tested. The weighing of the percentages shown based on subsystem fidelity and the flight envelope flown is presented in the analysis under each subsystem as identified by the WBS. Exhibit 14, page II-24, presents the flight envelopes of the RS-70 and the XB-70. The maximum allowable speed shown for the XB-70 was essentially 80 percent of the RS-70 maximum "Q" with a plateau at 30,000 ft. to reduce the impact of gust loading. The RS-70 was  $\pm$  2 "g" aircraft designed to also withstand gust loading of + 2 "g's", while the XB-70 was limited to + 1.6 "g's" and + 0.4 "g's" in maneuvering flight. Except for total temperature, the XB-70 was essentially limited to 80 percent of the RS-70 maximum allowable loading.

### SST FLIGHT RESEARCH

During the XB-70 flight test development program, special instrumentation was installed to obtain supersonic transport flight research data. The instrumentation was designed, manufactured, and installed (mainly on air vehicle no. 2) with data gathering, reduction, and analysis performed on a non-interference basis during the XB-70 development program. The SST flight research data was obtained in various areas as designated by the eight tasks shown below. All of the tasks indicated were sponsored by NASA except for acoustic loads which was sponsored by FAA.



	TASKS	A/V #1	<u>A/V #2</u>
1.	Aerodynamics (Panel Response)	x	x
2.	Structures (Landing Loads)	х	
3.	Structures (Gust Loads)		Х
4.	Structures (Acoustic Loads)		х
5.	Aerodynamics (Skin Friction Drag)		Х
6.	Aerodynamics (Base Drag)		Х
7.	Thermal Environment (Cabin & Structures	s)	Х
8.	Thermal Environment (Fuel System)	•	х

The instrumentation installed in the XB-70's for the SST flight research program is summarized as follows: (See Exhibit 15, page II-25.)

Panel Response	-	A NASA tape recorder and two microphones in the
		cockpit area.
Landing Loads	-	Rate of sink trailing arms, landing gear cameras, and NASA VGH recorder.
Gust Loads	-	A gust probe boom, accelerometers, pressure trans- ducers, and yaw and pitch angle instrumentation.
Acoustic Loads	-	Microphones and pressure tape to measure the properties of a turbulent boundary layer.
Skin Friction Drag	-	Wing pressure rakes, static pressure lines, thermo couple wires, and fuselage pressure rakes.
Base Drag	-	Boattail and pressure tapes around engines $4$ , 5, and 6.
Thermal Environment	-	Thermo couples on cabin windshield and structure, flight control components, landing gear, and fuel system.
Additional Data	-	The VGH recorder was also utilized to obtain fatigue data. The XB-70 development instrumentation, such as aerodynamic and stability and control transducers, were used for program correlation.

#### FOLLOW-ON RESEARCH ACTIVITY

Subsequent to the XB-70 flight test development program, two flight research programs were initiated on the remaining XB-70 No. 1. The first program was for Sonic Boom Measurements and was sponsored jointly by NASA and USAF. The first flight (No. 1-50) was initiated on 11-3-66 with the program completed on 1-31-67 (flight No. 1-60). The primary purpose of the test program was to determine the proper method of combining the theoretical sonic boom intensity due to lift and due to volume for the far field case. The pressure signatures of several aircraft, but principally the XB-70, were measured on the ground at various distances from the ground track of the air vehicle. The atmospheric effects on the sonic boom intensity were also investigated.

The second follow-on program was sponsored by NASA and was an investigation into the control of structural dynamics. The first flight (No. 1-61) was initiated on 4-25-67 with the program completed on 2-4-69 (flight No. 1-83)

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when the No. 1 XB-70 was ferried to WPAF Base Air Museum. To conduct the structure dynamics investigation, the air vehicle was modified for the installation of an Exciter Vane System in the nose section and an elevon control system titled Identically Located Accelerometers and Forces (ILAF).

The Exciter Vane System consisted of two small vanes protruding from both sides of the nose section, tied together through the fuselage by a "sewer pipe," and hydraulically driven. The system was controllable from the cockpit for both frequency and amplitude which provided controlled dynamics to the air vehicle structure. The ILAF system tied into the XB-70 Flight Augmentation Control System which provided elevon control for structural dynamic dampening. The concept was based on locating the input accelerometers near the elevons (for system stability), mixing this input signal with CG signals which cancelled normal flight accelerations and provided a structural dynamics frequency spectrum.

#### AIR VEHICLES 1, 2 and 3 CONFIGURATION COMPARISON

At the time of program redirection to three XB-70 air vehicles (3-31-61), it was planned that AV's 1 and 2 would be identical in design to demonstrate the technical feasibility of B-70 type aircraft design and AV 3 would include provisions to demonstrate the functional operation of a prototype bombing-navigation system. Subsequent to this redirection, systems development and manufacturing considerations led to further significant differences between AV's 1 and 2, as follows:

1. AV 1 had zero degree wing dihedral versus 5 degree on AV 2.

This change was issued to correct lateral dynamic stability problems at intermediate supersonic speed and became evident too late in the manufacturing process to be incorporated on AV 1 which utilized a bob weight to provide the capability for pilot compensation.

2. A manual air induction control system was utilized on AV 1 versus an automatic system on AV 2.

An interim manual system with limited automatic features was flown on AV 1 due to development schedules associated with the automatic system. See WBS 1.5 for additional detail.

3. The wing-to-fuselage joint design differed between AV's 1 and 2.

Due to wing-to-fuselage mismatch problems encountered on AV 1, the joining transition area was redesigned to facilitate assembly. See page II-26 for display of different design and WBS 3.0 for detail on problems encountered.

4. AV 2 had a 2800 gallon greater fuel capacity than AV 1.

This fuel capacity difference was primarily attributable to Tank No. 5 being blocked off on AV 1 due to problems experienced in sealing. See WBS 3.0 for further detail.



It was planned that AV 3 would be similar in structure as AV 2 but that modification would be made to include provisions for the bombing-navigation system as follows:

1. Four-man crew on AV 3 versus two-man crew on AV's 1 and 2.

An observer and system operator position were required in support of bombing-navigation missions and systems operations. See page II-27 for display of crew arrangement.

2. AV 3 required major modifications in the environmental control system.

Double versus single air cooling loops were required to support the additional crewmen and electronic equipment on AV 3. In addition, a greater cooling capacity and added liquid cooling loops were required. See Exhibit 18, page II-28.

3. Equipment bay variances

Major electronic equipment additions were required in support of the bombing-navigation system which resulted in rearrangement of the electronic and ECS equipment bays plus the blocking off of the No. 1 fuel tank for ECS water/ice tank.

4. Nose cavity additions

The AV 3 bombing-navigation required the addition of an inertial navigation platform and radar antenna equipment with associated cooling loops in the nose section. See Exhibit 17, page II-27.

5. Secondary power system additions

An additional generator on engine No. 5, line contactor, stepdown transformer, secondary bus and the associated controls were required in support of the added electronic equipment on AV 3.

#### MOCKUPS AND SIMULATORS

Shortly after contract go-ahead, effort was initiated on mockup design and fabrication in support of the Design Engineering Inspection (DEI). Emphasis was placed on the air vehicle mockup and displays, see Exhibit 19, page II-29. Additional mockups such as the over-nose vision (see Exhibit 20 page II-30), cabin lighting, and the Wright Air Development Center unit were being fabricated during this same period (See special chart, page II-31, for display of mockup milestones). Earlier program mockup requirements included those items associated with the weapons systems and mission; that is,

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alert pod, bomb displacing gear, rocket package, serial refueling, etc. As of July 1959, 21 individual major mockups and simulators had been identified. At the time of redirection to a 3 air vehicle program (March 1961) this number had increased to 24 and was as follows:

Flight Control Elect. Integration Rain Erosion Test Sled Doppler Radome Mockup ECM Radomes and Antennas Impedance Model Full Scale Vert. Stabilizer Antenna Range Models Radar Back Scatter Models Radar Reflectivity Environmental Propulsion Test Stand Engine Compartment Fire Tolerance Escape Capsule Test Sled Escape Capsule Test Units Electronic Systems Checkout Main Landing Gear Nose Landing Gear Hydraulic Pumping Antenna Test Mockup Bomb/Navigation Radome Mockup Engine-Inlet Compatibility Fuel System Fuel Tank Purge and Vent System

Due to the budget limitations and reduced program requirements associated with the 3 XB-70 air vehicle program, the aforementioned 24 items were reduced to the following ll:

Flight Control Doppler Radome Mockup Impedance Model .57 Scale Engine-Inlet Compat. Antenna Range Models Environmental Propulsion Test Stand Escape Capsule Test Units Electronic Systems Checkout Bomb/Navigation Radome Fuel System

The above units were completed in support of the XB-70 programs and further detail, if historically significant, appears under the individual subsystem in Volumes III and IV of this study.

#### COST DEFINITION

Cost data accumulated in this level 4 WBS item includes all the identifiable expenditures incurred in the design, development, ground testing, fabrication, assembly, system installation, vehicle checkout and preflight activities. In all cases, with the exception of system installation vehicle checkout and preflight, the costs are identified to a vehicle subsystem. Total costs of \$695,313,561 are segregated amoung the level 5 subsystems in the following manner:

WBS		Total Cost	Detail
1.1	Airframe Structure	\$406,036,084	Vol. III page III-65
1.2	Environmental Control Subsystem	24,906,411	Vol. III page III-404
1.3	Propulsion Subsystem	35,843,291	Vol. III page III-546
1.4	Secondary Power Subsystem	54,383,954	Vol. III page III-697
1.5	Air Induction Subsystem	19,060,408	Vol. IV page IV-65
1.6	Flight Control Subsystem	24,435,028	Vol. IV page IV-180

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WBS		Total Cost	<u>Detail</u>
1.7	Personnel Accom. & Escape		
_	Subsystem	\$ 12,557,555	Vol. IV Page IV-229
1.8	Alighting and Arresting Subsys.	17,259,216	Vol. IV Page IV-426
1.9	Mission and Traffic Control		
	Subsystem	3,770,766	Vol. IV Page TV-480
1.10	Flight Indication Subsystem	3.292.562	Vol. TV Page TV-574
1.11	Test Instrumentation Subsystem	8,937,176	Vol. TV Page TV-618
1.12	Subsystem Installation. Checkou	t -	
	and Preflight	61,338,601	Vol. TV Page TV-648
1.0	Air Vehicle *	23,492,509	
	Total 1.0 Air Vehicle	\$695,313,561	

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All cost data for WBS items 1.1 through 1.12 can be located in volume III or IV beginning on the pages indicated above. The remainder of this section will limit itself to the definition of the cost data associated at level 1.0 (\$23,492,509).

Detail of the recorded costs associated with this item is provided by Element of Cost (EOC) and Subdivision of Work (SDW). Section III of Volume I contains a definition of these items.

As an aid in the definition and evaluation of the in-house engineering costs associated with this WBS items, a matrix of the engineering hours expended by group has been developed. This matrix is as follows:

Group	Title	Hours
3	Electronical and Avionics Installation	63,051
11	Weight Control	259,843
12	Checking	6,829
13	Aerodynamics	57,726
14	Wind Tunnel Models	246,723
47	Human Factors and Cockpit Display	14,072
49	Avionics Integration and Control	73,692
51	Structural Dynamics	421,171
57	Engineering Specifications	40,819
64	Design Support	149,724
86	Electronic Integration	120,430
88	Electronic Equipment	58,619
94	Flight Simulation	7,291
95	Electrical System Design	9,286
97	Laboratory Services	36,764
109	Hydraulic Lab	10,296
110	Electrical Power Lab	95,149

\* Costs associated to the air vehicle as an entity and not identifiable to a particular subsystem.



		WBS CODE: 1.0
Group	Title	Hours
131 132 133 146	Aerodynamics Special Projects Thermodynamics Aerodynamics Thermodynamics Lab Various	5,078 17,873 54,456 5,582 72,105
	Total Engineering Hours	1,826,579
	1.0 Design Engineering SDW	1,745,000 hrs
	1.0 Test/QC SDW	81,579 hrs

1,826,579 hrs.

Ground testing activities identified to this item contain those test activities which could not be associated to a particular subsystem. These costs reflect the in-house testing activities only. Subcontractor testing, if identified, can be located in the Test/QC Subdivision of Work, Subcontracting Element of Cost, in each of the subsystems. The following is a summary of the major in-house test activities assigned to this WBS item.

Description	Recorded Cost
Air Vehicle Mock-up Model Shop-General Effort Metallic Materials Lab Tests Test Support Electrical Equipment Lab Tests Test Engineering Support Model Design - Construction General Test Shop Support Various	\$ 3,141,439 351,942 133,585 122,159 104,114 75,392 58,582 27,834 967,015
Cost (less MPC & G&A)	\$ 4,982,062
Material Procurement Cost	100,125
General and Administrative	42,888
Total Cost WBS 1.0 Test/QC SDW	\$ 5,125,075



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EXHIBIT 3

## HIGH-TEMPERATURE DEVELOPMENT

PRIOR TO B-70	B-70 DESIGN
240 F	450 TO 630 F
JP-4 $\sim$ 200 F	JP-6~300 F
300 TO 350 F	UP TO 450 F
250 F	550 F
-65 TO 350 F	-65 TO 450 F
375 F	630 F
275 F	500 TO 630 F
375 F (25 HR)	450 F (100 HR)
200 F (4 HR)	360 F (4 HR)
250 TO 300 F	500 TO 530 F
	PRIOR TO B-70 240 F JP-4 ~ 200 F 300 TO 350 F 250 F -65 TO 350 F 375 F 275 F 375 F (25 HR) 200 F (4 HR) 250 TO 300 F

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11-14

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## **RS-70 GENERAL ARRANGEMENT** 6 ENGINES INTERMEDIATE FUSELAGE J93-3 ∼ II STAGE ENGINE AIR DUCTS MULTI-WEAPONS BAY LANDING GEAR TWIN INTEGRAL FUEL TANKS VERTICALS ELEVONS ELECTRONIC EQUIP'T BAY -CREW OF 4 DELTA WING INTEGRAL FUEL AIR REFUEL FOLDING WING TIPS-**'RAM' ON INLET FACES INTEGRAL FUEL** NAV-LIGHTS - ECS CONDITIONING EQUIP'T BAY FLYING CANARD NOSE SECTION RADAR-SIGHTING ANTENNA INERTIAL NAV-PLATFORM DOPPLER RADAR

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SD72-SH-0003

EXHIBIT 5



RS-70 CATEGORY I  $\sim$  FLIGHT HOUR SUMMARY RS-70 TEST AIR VEHICLES



TASK	FLIGHT	HR										
2	AIR VEHICLE	50										
6	PROPULSION	315										
8	AIR FRAME	575										
9	ECS	100										
11	FLIGHT CONTROLS											
13	SECONDARY POWER	35										
PRIMAR	Y A/V TESTS: TOTAL	1240										
5	M&TC	115										
24	OFFENSIVE	400										
25	DEFENSIVE	250										
MILITAR	Y SUBSYSTEMS: TOTAL	<b>76</b> 5										
4	MISSILES	160	-									
26	PENETRATION	<u>110</u>										
AIRBORN	NE COOPERATION: TOTAL	270										
RS-70's	FLIGHT TEST HRS: TOTAL	2275										

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EXHIBIT 7

# RS-70 CATEGORY I $\sim$ FLIGHT HOUR SUMMARY FLYING TEST BEDS



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TASKS	<u>TITLE</u>										
6	PROPULSION	270									
14	4 PERSONNEL ACCOM										
PRIMARY	335										
4	MISSILES	150									
26	PENETRATION	100									
ASKS       IIILL         6       PROPULSION         14       PERSONNEL ACCOM         PRIMARY A/V TESTS: TOTAL       4         4       MISSILES         26       PENETRATION         AIRBORNE COOPERATION: TOTAL         24       OFFENSIVE         25       DEFENSIVE         MILITARY SUBSYSTEMS: TOTAL         FLYING TEST BEDS FLIGHT HR: TOTAL		250									
24	OFFENSIVE	1925									
25	DEFENSIVE	150									
MILITARY	SUBSYSTEMS: TOTAL	2075									
FLYING	TEST BEDS FLIGHT HR: TOTAL	2660									

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. RS-70 CATEGORY I ~ FLIGHT HOUR SUMMARY CHARGEABLE FLIGHT HOURS



TASKS	TITLE	<u>FI</u>	<u>IGHT H</u>	<u>R</u>
2	AIR VEHICLE		50	
6	PROPULSION		435	
7	AIR INDUCTION SYS		150	
. 8	AIRFRAME		575	
	STABILIZE AND CONTROL	250		
	PERFORMANCE	140		
	STRUCTURES	85		
	a/v demo	100		
	ALIGHT AND ARREST	0		
9	ECS		100	
11	FLIGHT CONTROL		165	
12	FLIGHT INDICATION		0	
13	SECONDARY POWER		35	
14	PERSONNEL ACCOMMODATIO	NS	65	
СНА	RGEABLE FLIGHT HR.: TOTAL		1575	

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EXHIBIT 9 SD72-SH-0003

## RS-70 CATEGORY I $\sim$ FLIGHT HOUR SUMMARY PRIMARY AIR VEHICLE EQUIVALENT FLIGHT TEST HOURS

		PERCENT FLIGHT HOURS ADDITIVE DUE TO TIME-SHARING													IGHT
PRIMARY AIR VEHICLE SYSTEMS	BASIC FLIGHT HOURS CHARGED PER SYSTEM	AIR VEHICLE	PROPULSION	AIR INDUCTION	STABILITY & CONT.	PERFORMANCE	STRUCTURES	A/V DEMO	ECS	FLIGHT CONT.	FLIGHT INDICATION	SECONDARY PWR	Personnel accom	NET TIME SHARING (H	TOTAL EQUIVALENT FL TEST HOURS
AIR VEHICLE	50		_	-	-	_	-	_	-	1	-	_	-	-	50
PROPULSION	435	-	-	50	10	50	-	10	10	ł	-	100	-	225	660
AIR INDUCTION	150	-	60	-	10	50	-	10	10	20	-	_	_	245	395
STABILITY AND CONTROL	250	-	10	10	-	25	50	10	-	60	-	_	-	220	470
PERFORMANCE	140	-	50	50	25	-	-	10	10	-	1	-	_	235	375
STRUCTURES	85	-	20	20	50	25	_	50	25	60	-		-	400	485
A/V DEMO	100	-	-	-	-	-	-	I	-	-	-		-	_	100
ALIGHT AND ARREST	0	-	20	20	20	20	20	20	-	-	-	_	-	180	180
ECS	100	-	10	10	10	10	50	20	-	10	1	-	-	180	280
FLIGHT CONTROL	165	-	10	10	75	10	25	10	-	-	-	-	_	260	425
FLIGHT INDICATION	0	-	10	10	50	25	25	10	10	50	-	-	-	310	310
SECONDARY POWER	35	-	100	20	10	10	10	10	10	10	_		-	295	330
PERSONNEL ACCOM.	65	-	.10	10	10	10	10	10	50	10	-	_	_	155	220
											TC	OTAL			4280

EXHIBIT 10 SD72-SH-0003 F





EXHIBIT 11

## XB-70 PROGRAM ~ FLIGHT HOUR SUMMARY PRIMARY AIR BEHICLE EQUIVALENT FLIGHT TEST HOURS

		PERCENT FLIGHT HOURS ADDITIVE DUE TO TIME-SHARING												OURS)	LIGHT
PRIMARY AIR VEHICLE SYSTEMS	BASIC FLIGHT HOURS CHARGED PER SYSTEM	AIR VEHICLE	PROPULSION	AIR INDUCTION	STABILITY & CONT.	Performance	STRUCTURES	ALIGHT & ARREST	ECS	FLIGHT CONT.	FLIGHT INDICATION	SECONDARY PWR	Personnel accom	NET TIME SHARING (H	TOTAL EQUIVALENT F TEST HOURS
AIR VEHICLE	20	-	-	_	I	-	. –	-	-	-		-,	-	<b>.</b>	20
PROPULSION	35	-	_	50	10	50	-	-	-	-	-	-	-	40	75
AIR INDUCTION	30	-	60	-	10	50	-	-	-	-	-	-	-	45	75
STABILITY AND CONTROL	50	-	10	10	-	25	50	-	I	-	-	-	-	20	70
PERFORMANCE	40	_	50	50	25	-	-	-	-	-	-	-	-	45	85
STRUCTURES	10	-	20	20	50	25	-	-	-	-	-		-	50	60
ALIGHT AND ARREST	0	1	20	20	20	20	20	-		-	-	-	-	35	35
ECS	0	-	10	10	10	10	50	-	-	-	-	_	-	20	20
FLIGHT CONTROL	0	_	10	10	75	10	25	-	-	-	-	-	-	50	50
FLIGHT INDICATION	0	-	10	10	50	25	25	-	-	-	-	-	-	45	45
SECONDARY POWER	0	-	100	20	10	10	10	-	-	-	-	-	-	50	50
PERSONNEL ACCOM	120	-	10	10	10	10	10	-	-	-	-	-	-	15	135
TOTAL													720		

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EXHIBIT 12 SD72-SH-0003

## XB-70 PROGRAM $\sim$ FLIGHT HOUR EXPOSURE BASED ON RS-70 PROGRAM TO BE UNITY



EXHIBIT 13

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SD72-SH-0003

EXHIBIT 15

## WING TO FUSELAGE JOINING



AIR VEHICLE 2



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EXHIBIT 16 SD72-SH-0003
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# BNEMG SUBSYSTEM INSTALLATION AIR VEHICLE NO 3



# EUVIRONMENTAL CONTROL SUBSYSTEM



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## ORIGINAL PAGE BLACK AND WHITE PHOTOGRAPH





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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS **B-70 AIRCRAFT STUDY**

#### 4-SYSTEM 1 AIR VEHICLE

			5-SUBSYS	5-SUBSYS	5-SUBSYS	5-SUBSYS
			HOURS	HOURS		
			DOLLARS	DOLLARS	DOLLARS	DOLLARS
DESIGN/ENGINEE	RING	18	26579	8302581	985802	1761024
LABOR AT \$	4.921		8505232	40769287	4912462	8489644
ENGR BURDEN	AT \$	4.463	7919655	36746727	4335751	7185430
PRODUCTION .			1	1904709		
LABOR AT \$	3.222		· · · · · · · · · · · · · · · · · · ·	38337731		
SHOP SUPPORT		. 5	08479	4386037	111236	442874
LABOR AT \$	3.036		1488671	13183710	320730	1393646
PLANN ING				716532	223720	10,0040
LABOR AT \$	3.417			2442730		
TEST/QC			5411	1950765	8931	27838
LABOR AT \$	3.531		17456	6871767	27132	87911
MFG BURDEN	AT \$	3.918	1966438	74073403	418209	1717820
ENGR MATERIAL			<b>916</b> 866	11693669	167696	815446
MFG MATERIAL				27227491		012440
SUBCONTRACT				103499447	13552727	11598442
MPC			106624	8205547	582839	608353
WIND TUNNEL				2760920		
OTHER COST			2267739	12295983	197805	3383427
SUB-TOTAL			23188681	378108412	24515351	35280119
GEN & ADMIN			303828	6595097	391060	563172
I DWA			•	21 33 25 75		
TOTAL COST			23492509	406036084	24906411	35843291
	SUBDI <b>VI</b> Cost de	SION OF WORK TAIL - SEE PAGE	II-36	III-73	III-438	TTT-500

III-73

III-438

III-590

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM '1 AIR VEHICLE

			5-SUBSYS	5-SUBS YS	5-SUBSYS	5-SUBSYS
			HOURS	HOURS	HOURS	HOURS
			DOLLARS	DOLLARS	DOLLARS	DOLLARS
DESIGN/ENGIN	IEERING		1950276	630695	1130155	526124
LABOR AT	\$ 4.921		9832880	3110162	5567013	2580264
ENGR BURDEN	AT \$	4.463	9231077	2700615	5118121	2375291
PRODUCT ION						
LABOR AT	\$ 3.222					
SHOP SUPPORT	. '		339739	151065	573357	510049
LABOR AT	\$ 3.036		1041422	470196	1712747	1610946
PLANN ING						
LABOR AT	\$ 3.417					
TEST/QC			19595	10036	34093	21266
LABOR AT	\$ 3.531		63051	31264	110134	65983
MFG BURDEN	AT \$	3.918	1354238	629614	2232702	1932869
ENGR MATERIA	L		805279	546226	1173454	966093
MFG MATERIAL						3
SUBCONTRACT			29005372	9301831	7425036	1153331
MPC			1369796	412883	427124	146407
WIND TUNNEL				1243178		
OTHER COST			330209	326113	257964	69926
SUB-TOTAL			53033324	18772082	24024295	10901110
GEN & ADMIN			921444	288326	410733	200741
IDWA			429186			1455704
TOTAL COST			54383954	19060408	24435028	12557555
	cine nt ut ci	OW OF HOPY				
	COST DET	AIL - SEE PAGE	ITI-734	<b>IV-</b> 94	IV-216	IV-336

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

#### 4-SYSTEM 1 AIR VEHICLE

			5-SUBSYS 08 HOURS DOLLARS	5–SUBSYS 09 HOURS DOLLARS	5-SUBSYS 10 HOURS DOLLARS	5-SUBSYS 11 HOURS DOLLARS
DESIGN/ENGINE LABOR AT \$ ENGR BURDEN	ERING 4•921 AT \$	4.463	199309 998488 874375	213308 995250 908594		530886 2686991 2647685
PRODUCTION LABOR AT \$ SHOP SUPPORT LABOR AT \$ PLANNING	3.222 3.036		2650 9280	16223 47473		137320 449311 174639 591227
LABOR AT \$ TEST/QC LABOR AT \$ MFG BURDEN	3.417 3.531 AT \$	3.918	301 1273 11602	771 2617 62860		19251 66972 1435666
ENGR MATERIAL MFG MATERIAL			8199	98457		747683
SUBCONTRACT MPC WIND TUNNEL			14352749 704276	1505370 85538	3097830 136403	104605
OTHER COST			61	9157		48942
SUB-TOTAL			16960303	3715316	3234233	8779082
GEN & ADMIN IDWA			298913	55450	58329	158094
TOTAL COST			17259216	3770766	3292562	8937176
	SUBDIV Cost d	ISION OF WORK ETAIL - SEE PA	<b>AGE</b> IV-447	IV-511	IV-577	I <b>V-</b> 620

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 1 AIR VEHICLE

			5-SUBSYS	
		1	12	TOTAL
			HOURS	HOURS
			DOLLARS	DOLLARS
DESIGN/ENGINI	EERING		316502 1	8373241
LABOR AT	\$ 4.921		1967649	90415322
ENGR BURDEN	AT \$	4.463	1952457	81995778
PRODUCTION			4590814 1	6632843
LABOR AT	\$ 3.222		14796835	53583877
SHOP SUPPORT			65703	7282051
LABOR AT	\$ 3.036		236401	22106449
PLANN ING			280324	996856
LABOR AT	\$ 3.417		963821	3406551
TEST/QC			587203	2685461
LABOR AT	\$ 3.531		2135631	9481191
MFG BURDEN	AT \$	3.918	22277981	108113402
ENGR MATERIAL	L			17939068
MFG MATERIAL			9848132	37075623
SUBCONTRACT				194492135
MPC			1318030	14208425
WIND TUNNEL		•		4004098
OTHER COST				19187326
SUB-TOTAL			55496937	656009245
GEN & ADMIN			1155919	11401106
IDWA			4685745	<b>2790321</b> 0
TOTAL COST			61338601	695313561
	SUBDIVISI	ON OF WORK		<b>TT</b> ( <b>0</b>
	COST DETA	ll - See Page	14-00-	TT-00

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 1 5-SUBSYSTEM 0

#### AIR VEHICLE

			DESIGN	TEST	
			/ENGR	/QC	TOTAL
			HOURS	HOURS	HOURS
			. DOLLARS	DOLLARS	DOLLARS
DESIGN/ENGINEER	ING		1745000	81579	1826579
LABOR AT \$	4.656		8158719	346513	8505232
ENGR BURDEN	AT \$	4.336	7550198	369457	7919655
SHOP SUPPORT			14412	494067	508479
LABCR AT \$	2.928		44856	1443815	1488671
TEST/QC			462	4949	5411
LABOR AT \$	3.226		1668	15788	17456
MFG BURDEN	AT \$	3.827	62908	1903530	1966438
ENGR MATERIAL			31391	825475	016966
MPC			6499	100125	1066 26
OTHER COST			2250255	17484	2267739
SUB-TOTAL			18106494	5082187	23188681
GEN & ADMIN			260940	<b>42</b> 888	303828
TOTAL COST			18367434	5125075	23492509

TIME-PHASED COST DETAIL - SEE PAGE II-37 II-47 II-52

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

DESIGN/ENGINEERING 4-SYSTEM 1 AIR VEHICLE 5-SUBSYSTEM 0 SUBD OF WORK DESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58	162.0	27270	4.574	124733	124080	248813
Q-2	58						
Q-3	58	707.5	118922	4.289	510012	463263	973275
Q-4	58						
0 - 1	59	826.0	140995	4.169	587867	481887	1069754
Q-2	59						r
Q-3	59	1083.0	190654	4.258	811889	698724	1510613
Q-4	59						
Q-1	60	1002.0	173710	4.561	792331	649651	1441982
Q-2	60						
Q-3	60	877.0	147300	4.800	707037	543459	1250496
Q-4	60						
Q-1	61	1303.0	222341	4.675	1039462	758136	1797598
Q-2	61						
Q-3	61	842.5	152750	4.918	751208	694903	1446111
Q-4	61						
Q-1	62	603.0	102930	4.967	511216	473422	984638
Q-2	62						
Q-3	62	435.5	73144	4.861	355588	371280	726868
Q-4	62						
Q-1	63	549.0	93733	5.027	471209	507067	978276
Q-2	63						
Q-3	63	607.0	101986	5.001	510006	581447	1091453
Q-4	63						
Q-1	64	948.0	161790	4.786	774294	966461	1740755
Q-2	64						
Q-3	64	40.0	7060	6.493	45842	48088	93930
Q-4	64						
Q-1	65	148.0	25598	5.537	141729	163863	305592
0-2	65						

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

DESIGN/ENGINEERING 4-SYSTEM 1 5-SUBSYSTEM 0 AIR VEHICLE SUBD OF WORK DESIGN/ENGINEERING

LABOR + BURDEN \$	BUR DEN Doll Ars	LABOR DOLLARS	LABOR RATE	LABOR HOURS	MAN- MONTHS	
22552	9008	13544	5.479	2472	14.5	Q-3 65 Q-4 65
26211	15459	10752	4.585	2345	13.5	Q-1 66
15708917	7550198	8158719		1745000	10161.5	TOTAL

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 1 AIR VEHICLE 5-SUBSYSTEM 0 SUBD OF WORK DESIGN/ENGINEERING

#### **ON-SITE LABOR**

		MAN- MON THS	LABOR HOUR S	LABOR	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58	10.5	1712	2.992	5123	4928	10051
Q-2	58						
Q-3	58		-77	1.351	-104	-193	-297
Q-4	58						
Q-1	59	1.5	260	3.546	922	962	1884
Q-2	59						
Q-3	59	24.0	4314	2 <b>.887</b>	12455	17410	29865
Q-4	59						
Q-1	60	-16.5	-2881	2.885	-8313	-7854	-16167
Q-2	<b>6</b> 0						
Q-3	60		98	2.653	260	216	476
Q-4	60						
Q-1	61		69	2 92.8	202	236	438
Q-2	61						
Q-3	61	3.0	636	2.943	1872	2260	4132
Q-4	61						
Q-1	62	12.0	1925	3.068	5905	7538	13443
Q-2	62						
Q-3	62	10.5	1847	3.001	5543	7339	12882
Q4	62						
0-1	63	13.5	2259	3.205	7240	9312	16552
Q-2	63				•		
0-3	63	12.0	1930	3.307	6383	9473	15856
Q-4	6.3						
Q-1	64	10.5	1788	3.147	5627	8747	14374
Q-2	64						
Q-3	64	3.0	515	3.266	1682	2411	4093
ų-4	64			<b>.</b>			
0-1	65		17	3.471	59	111	170
Q-2	65						

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### SHOP SUPPORT

4-SYSTEM1AIR VEHICLE5-SUBSYSTEM0SUBD OF WORK DESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 65					12	12
TOTAL	84.0	14412		44856	62908	107764

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### TEST/QC 4-SYSTEM 1 AIR VEHICLE 5-SUBSYSTEM 0 SUBD OF WORK DESIGN/ENGINEERING

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58		49	3.061	150		150
Q-2	58						
Q-3	58		76	3.039	231		231
Q-4	58						
Q-1	59						
Q-2	59						
Q-3	59		115	4.626	532		532
Q-4	59						
Q-1	60	1.5	149	3.322	495		495
Q-2	60						
Q-3	60		13	2.846	37		37
Q-4	60						
Q-1	61						
Q- 2	61						
Q-3	61						
Q 4	61						
Q-1	62						
Q-2	62						
Q-3	62		6	4.667	28		28
Q-4	62						
Q-1	63		35	3.029	106		106
Q-2	63				• •		
Q-3	63						
Q-4	63						·
Q-1	64						
Q-2	64						1
Q-3	64	· ·	9	5.111	46		46
0-4	64					•	
Q-1	65		6	4.833	29		29
0-2	65						

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### TEST/QC

4-SYSTEM1AIR VEHICLE5-SUBSYSTEM0SUBDOF WORKDESIGN/ENGINEERING

	£.	HAN- MONTHS	LABOR HOURS	LABUR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 65			4	3.500	14		14
TOTAL		1.5	462		1668		1668

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM	1	ATR	VENTCLE	
5-SUB SYSTEM	0			
SUBD OF WORK	DESIGN/ENGINEERIN	1G		

		MAN-	LABOR	LABOR	LABOR	BUR DEN	LABOR +	ENGR
		MONTHS	HOURS	RATE	DOLL AR S	DOLL ARS	BURDEN \$	MATL
Q-1	58	172.5	29031	4.478	130006	129008	259014	580
Q-2	58							
Q-3	58	707.5	118921	4.290	510139	463070	973209	-174
Q-4	58							
Q-1	59	827.5	141255	4.168	588789	482849	1071638	6
Q-2	59							
Q-3	59	1107.0	195083	4.228	824876	716134	1541010	2391
Q-4	59							
Q - 1	60	987.0	170978	4.588	784513	641797	1426310	-386
0-2	60							
0-3	60	877.0	147411	4.798	707334	543675	1251009	119
0-4	60							
$\tilde{0}-1$	61	1303-0	222410	4.675	1039664	758372	1798036	97
0 - 2	61	200.000			2007001		1190000	~ 1
0 - 3	61	845.5	153386	4-910	753080	697163	1450243	200
0-4	61	0.202	100000		122000	071105	1120212	200
0-1	62	615.0	104855	4.932	517121	480960	998081	2000
0-2	62	UI JOU	1040))	70 332	711121	400 300	770001	2155
0-2	62	446 0	74007	6 916	261150	279610	720770	2726
Q-5 0-4	42	440.0	14771	7.010	501159	210013	137110	212.0
Q-4 0 1	62	540 S	0(007	4 004	170555	51/ 370	004024	173
Q-1 0 7	(7)	202+2	90021	4. 704	410222	210213	774754	4/3
Q-2	0.3	(10.0	102017	1 0 0	51 ( 30 0	500.000	1107000	<b>67</b> 0
Q-3	63	019.0	103410	4.909	210387	590920	110/309	550
Q-4	03	650 <b>5</b>			330001	075000	1755100	1010
Q-1	64	958.5	103578	4. 168	779921	975208	1755129	1210
Q-2	64							
Q-3	64	43.0	7584	6.272	47570	50499	98069	1053
Q-4	64							
0-1	65	148.0	25621	5.535	141817	163974	305791	14477
Q-2	65							
Q-3	65	14.5	2476	5.476	13558	9020	22578	4120
Q-4	65							

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	4-SYSTEM 5-SUBSYSTEM		1 0	AIR VE	HICLE				
	SUBD	BD OF WORK DESIGN/ENGINEERING							
		MAN- MON TH S	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR Matl	
Q-1 Q-2 Q-3	66 66 66	13.5	2345	<b>4.5</b> 85	10752	15459	26211	-1 951	
TOT	ΓAL	10247.0	1759874		8205243	7613106	15818349	31391	

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

4-SYSTEM 1 AIR VEHICLE 5-SUBSYSTEM 0 SUBD OF WORK DESIGN/ENGINEERING

		MPC	OTHER COST	SUB Total	G&A	TOTAL COST
0 - 1 0 - 2	58 58	31		259625		259625
Q-3	58 58	-9		973026		9 <b>7</b> 3026
$\overline{Q} - 1$	59 ···	1	34402	1106047		1106047
Q-2 Q-3	59 59	202	92803	1636406		1636406
$Q^{-1}$ $Q^{-1}$	60 60	50	124088	1550062	29533	1579595
Q=2 Q=3	60 60	15	188561	1439704	27431	1467135
Q-1	61 61	8	332312	2130453	<b>3959</b> 0	2170043
Q-2 Q-3	61	16	200337	1650796	30677	1681473
Q=4 Q=1 Q=2	62 62	236	381992	1383308	23219	1406527
Q=2 Q=3	62 62	214	303001	1045719	17552	1063271
Q-4 Q-1	63 63	46	166291	1161744	19424	1181168
Q-3	63 63	54	-89817	1018096	17023	1035119
Q - 4 Q - 1 Q - 2	65 64	128	154695	1911162	40666	1951828
Q = 2 Q = 3	64 64	383	153568	253073	5385	258458
Q-1 Q-2	65 45	4330	145615	470213	7209	477422
Q-3 0-4	65 65	735	58246	85679	2286	87965
Q-4	65					

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

TOTAL

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	4-SYSTEM 5-SUBSYSTEM		1 0	AIR VEHICLE							
	SUBD OF WORK		DES	DESIGN/ENGINEERING							
			MP(	2	OTHER CDST	s Tot	SUB Fal	G	3	A	TOTAL COST
Q-1	66 66		1	L	4161	303	372		91	5	31287
Q-3	<b>6</b> 5		58	Ĵ		10	009		- 3	0	1039

6499 2250255 18106494 260940 18367434

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESIGN/ENGINEER	ING	
4-SYSTEM	1		
5-SUB SYSTEM	0		
SUBD OF WORK	TEST/QC	AIR	VEHICLE

		MAN- MONTHS	LABOR HOURS	LABOR	LABOR DOLLAR S	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-3	60	40.5	6752	4.720	31872	25055	56927
Q-4	60						
Q-1	61	37.5	6426	3.914	25153	21891	47044
Q-2	61						
Q-3	61	145.5	26415	3.914	103398	106238	209636
Q-4	61						
Q-1	62	135.0	23098	3.964	91550	105335	196885
Q-2	62						
0-3	62	30.0	4953	4.663	23098	29173	52271
Q-4	62						
Q-1	63	15.0	2654	4.340	11519	13331	24850
Q-2	63						
Q-3	63	33.0	5466	5.368	29344	30972	60316
Q-4	63						
Q-1	64	34.5	5815	5.259	30579	34726	65305
Q-2	64						
Q-3	64					2736	2736
тот	AL	471.0	81579		346513	369457	715970

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### SHOP SUPPORT 4-SYSTEM 1 5-SUBSYSTEM 0 SUBD OF WORK TEST/QC

## AIR VEHICLE

		MAN-	LABOR	LABOR	LABOR	BUR DEN	LABOR +
		MUNTHS	HOURS	RATE	DOLLARS	DOLL ARS	BURDEN \$
0-3	58	3.0	625	2.765	1728	1893	3621
Q-4	58						
Q-1	59	1647.0	281018	2.927	822640	1008290	1830930
Q-2	59						
Q-3	59	516.0	90943	2.810	255535	404368	659903
Q-4	59						
Q-1	60	420.0	72852	2.871	209123	279654	488777
Q-2	<b>6</b> 0						
Q-3	<b>6</b> 0	7.5	1328	3.541	4703	-9136	-4433
Q-4	60						
Q-1	61	52.5	9066	3.144	28502	31115	59617
Q-2	61						
Q-3	61	16.5	2881	3.420	9854	11986	21840
Q-4	61						
Q-1	62	6.0	1020	3.438	3507	5041	8548
Q-2	62						
Q-3	62	1.5	132	3.364	444	646	1090
Q-4	62						
Q-1	63	1.5	274	3.620	992	1115	2107
Q-2	63						
Q-3	63	76.5	12901	3.030	39086	58009	97095
Q-4	63				a.		
0-1	64	123.0	21027	3.220	67701	111529	179230
Q-2	64						·
Q-3	64					-980	-980
тот	AL.	2871.0	494067		1443815	1903530	3347345

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	TEST/QC	
4-SYS TEM	1	
5-SUB SYSTEM	0	AIR VEHICLE
SUBD OF WORK	TEST/QC	

		MAN- MON TH S	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	<b>5</b> 8		5	3.000	15		15
Q-4	53						
Q-1	59	7.5	1362	3.148	4287		4287
Q-2	59						
Q-3	59	3.0	474	2.987	1416		1416
Q-4	59						
Q-1	60	1.5	201	3.189	641		641
Q-2	60						
Q-3	<b>6</b> 0	1.5	151	4.576	691		691
Q-4	60	,					,
Q-1	61		52	2.404	125		125
Q-2	61						
Q-3	61		37	4.811	178		178
Q-4	61						
Q-1	62		9	2.889	26		26
Q-2	62						
Q-3	62						
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	63	4.5	746	2.964	2.211		2211
Q-4	63						
Q-1	64	10.5	1912	3.242	5198		6198
Q-2	64						
Q-3	64						
тот	AL	28.5	4949		15788		15788

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

AIR VEHICLE

4-SYSTEM 1 5-SUBSYSTEM 0 SUBD OF WORK TEST/QC

		MAN- MON THS	LABOR HOUR S	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
Q-3 Q-4	58 58	3.0	630	2.767	1743	1893	3636	4207
Q-1 Q-2	59 59	1654.5	282380	2.928	826927	1008290	1835217	212172
Q-3 Q-4	59 59	519.0	91417	2.811	256951	404368	661319	37647
Q-1 Q-2	60 60	421.5	73053	2.871	209764	279654	489418	31574
Q-3 Q-4	60 60	49.5	8231	4.528	37266	15919	53185	2230
Q-1 Q-2	61 61	90.0	15544	3.460	53780	53006	106786	17296
Q-3 Q-4	61 61	162.0	29333	3.867	113430	118224	231654	5821
Q-1 Q-2	62 62	141.0	24127	3.941	95083	110376	205459	91483
Q-3 Q-4	62 62	31.5	5085	4.630	23542	29819	53361	173178
0-1 0-2	63 63	16.5	2928	4.273	12511	14446	26957	27529
Q-3 Q-4	63 63	114.0	19113	3.696	70641	88981	159622	55987
Q-1 Q-2	64 64	168.0	28754	3.634	104478	146255	250733	151924
Q-3	64					1756	1756	74427
тот	AL	3370.5	580595		1806116	2272987	4079103	885475

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

> > AIR VEHICLE

4-SYSTEM 1 5-SUBSYSTEM 0 SUBD DF WORK TEST/QC

		MPC	OTHER COST	SUB Total	GεA	TOTAL COST
Q-3	58	230		8073		8073
Q-4	58					
Q-1	59	17965	4967	2070321		2070321
0-2	59					
Q-3	59	3187	3571	705724		705724
Q-4	59				·	
0 - 1	60	4153		525145	10006	535151
Q-2	60					
Q-3	60	293	2292	58000	1105	59105
Q-4	60					
Q-1	61	1460	3328	128870	2395	131265
Q-2	61					
Q-3	61	491	2591	240557	4470	245027
Q-4	61					
Q-1	62	7205	737	304884	5117	310001
Q-2	62					
Q-3	62	13639	10289	250467	4204	254671
Q-4	62				`	
Q-1	63	2712	19093	76291	1276	77567
Q-2	63					
Q-3	63	5516	-29384	191741	3206	194947
Q-4	63					
Q-1	64	16194		418851	8912	427763
0-2	64			•		
Q-3	64	27080		103263	2197	105460
TOT	AL	100125	17484	5082187	42888	5125075

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESIGN/ENGINEERI	NG	
4-SYSTEM	1		
5-SUB SYSTEM	0	AIR	VEHICLE
		,	

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q - 1 Q - 2	58 58	162.0	27270	4.574	124733	124080	248813
Q-3 Q-4	58 58	707.5	118922	4.289	510012	463263	973275
Q-1 Q-2	59 59	826.0	140995	4.169	587867	481887	1069754
Q-3 Q-4	-59 59	1083.0	190654	4.258	811889	698724	1510613
Q-1 Q-2	60 60	1002.0	173710	4.561	792331	649651	1441982
Q-3 Q-4	60 60	917.5	154052	4.796	738909	568514	1307423
Q-1 Q-2	61 61	1340.5	228767	4.654	1064615	780027	1844642
Q-3 Q-4	61 61	988.0	179165	4.770	854606	801141	1655747
Q-1 Q-2	62 62	739.0	126028	4.783	602766	578757	1181523
Q-3 Q-4	62 62	464.5	78097	4.849	378686	400453	779139
Q-1 Q-2	63 63	565.0	96387	5.008	482728	520398	1003126
Q-3 Q-4	63 63	640.0	107452	5.019	539350	612419	1151769
Q-1 Q-2	64 64	982.0	167605	4.802	804873	1001187	1806060
Q-3 Q-4	64 64	40.0	7060	6.493	45842	50824	96666
Q-1 Q-2	65 65	148.0	25598	5.537	141729	163863	305592
Q-3	65	14.5	2472	5.479	13544	9008	22552

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESI	GN/ENGINEERING	
4-SYSTEM	1		
5-SUB SYSTEM	0	AIR	VEHICLE

LABOR + BURDEN \$	BUR DEN DOLL AR S	LABOR DOLLARS	LABOR Rate	LABOR HOURS	MAN- MONTHS	
26211	15459	10752	4.585	2345	13.5	Q-4 65 Q-1 66
16424887	7919655	8505232		1826579	10633.0	TOTAL

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 5-SUBSYSTEM	SHOP 1 0	SUPPORT	AIR	VEHICLE		
	ON-S	ITE LABOR				
MAN	<i>i</i> -	LABOR	LABOR		IR	BURDEN

		MAN- MON TH S	LABOR HDURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL AR S	LABOR + BURDEN \$
Q-1 Q-2	58 58	10.5	1712	2.992	5123	4928	10051
Q-3 Q-4	58 58	3.0	548	2.964	1624	1700	3324
Q-1 Q-2	59 59	1648.5	281278	2.928	823562	1009252	1832814
Q-3 Q-4	59 59	541.5	9525 <b>7</b>	2.813	267990	421778	689768
Q-1 Q-2	60 60	403.5	69971	2.870	200810	271800	472610
Q-3 Q-4	60 60	9.0	1426	3.480	4963	-8920	-3957
Q-1 Q-2	61 61	54.0	9135	3.142	28704	31351	60055
Q-3 Q-4	61 61	19.5	3517	3.334	11726	14246	25972
Q-1 Q-2	62 62	17.5	2945	3.196	9412	12579	21991
Q-3 Q-4	62 62	12.0	1979	3.025	5987	7985	13972
Q-1 Q-2	63 63	15.0	2533	3.25.0	8232	10427	18659
Q-3 Q-4	63 63	88.5	14831	3.066	45469	67482	112951
Q-1 Q-2	64 64	133.5	22815	3.214	73328	120276	193604
Q-3 Q-4	64 64	3.0	515	3.266	1682	1431	3113
Q-1 Q-2	65 65		17	3.471	59	111	170
Q-3	65					12	12
тот	AL	2959.0	508479		1488671	1966438	3455109

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> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## TEST/QC

4-SYSTEM 1 5-SUBSYSTEM 0

#### AIR VEHICLE

#### ON-SITE LABOR

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL AR S	LABOR + BURDEN \$
0-1	58		49	3,061	150		150
$\tilde{0}-\tilde{2}$	58			5.001	120		170
Q-3	58		81	3.037	246		246
Q-4	58		_				
Q-1	59	7.5	1362	3.148	4287		4287
0-2	59						
Q-3	59	3.0	589	3.307	1948		1948
Q-4	59						
Q-1	60	1.5	350	3.246	1136		1136
Q-2	60						
Q-3	60	1.5	164	4.439	728		728
Q-4	60						
Q-1	61		52	2.404	125		125
Q-2	61						
Q-3	61		37	4.811	178		178
Q-4	61						
Q-1	62		9	2.889	26		26
Q-2	62						
Q-3	62		6	4.667	28		28
Q-4	62		_				
Q-1	63		35	3.029	106		106
Q-2	63	<i>.</i> _					
Q-3	63	4.5	746	2.964	2211		2211
0-4	63	10 5	1010	2 24 2	(100		( ) 0 0
Q = 1	64	10.5	1912	3.242	6198		6198
Q=2	64		•	6 111			
Q-3	64	•	9	2.111	40		40
Q-4	64		(	( 027	20		20
$\psi = 1$	60		0	4.833	23		29
<b>∀</b> <sup></sup> ∠	00 65			3 500	1 /		1.4
Q-0	09		4	5.500	14		14
тот	AL	28.5	5411		17456		17456

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 5-SUBSYSTEM 0

#### AIR VEHICLE

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
0-1	EO	170 5	20001					
Q-1 Q-2	20 58	112.5	29031	4.4/8	130006	129008	259014	580
Q-3	58	710.5	119551	4.282	511882	464963	976845	4033
Q-4	58					101705	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	1035
Q-1	59	2482.0	423635	3.342	1415716	1491139	2906855	212178
Q-2	<b>5</b> 9							
Q-3	59	1627.5	286500	3.776	1081827	1120502	2202329	40038
0 - 1	60	1407.0	244031	4 074	99/ 277	021751	1015720	71100
$\hat{\mathbf{Q}} = \hat{\mathbf{Z}}$	60	140100	244031	4.014	334211	921491	1910120	21100
Q-3	60	928.0	155642	4.784	744600	559594	1304194	2349
Q-4	<b>6</b> 0							
Q-1	61	1394.5	237954	4.595	1093444	811378	1904822	17393
Q-2	61	1007 5	100710					
Q-3	61	1007.5	182719	4.142	866510	815387	1681897	6021
0-1	62	756.5	128982	4.746	612204	501336	1203540	044.92
$\tilde{Q}-2$	62	1900	120,02	4.140	01221.4	271,220	1203340	34402
Q-3	62	476.5	80082	4.804	384701	408438	793139	175904
Q-4	62							
Q-1	63	580.0	98955	4.963	491066	530825	1021891	28002
Q-2	63	722.0	100000					
0-4	63	435.0	123029	4.1/1	587030	67990I	1266931	56537
0-1	64	1126.0	192332	4.598	884399	1121463	2005862	153134
Q-2	64			10,270	001977	1121405	2009002	179194
Q-3	64	43.0	7584	6.272	47570	52255	99825	75480
Q-4	64							
Q-1	65	148.0	25621	5.535	141817	163974	305791	14477
Q-2	65 45	14 5	2/7/	E /7/	12550		<b>5353</b> 0	
Q-3 0-4	00 65	14.7	2410	2+475	13228	9020	22518	4120
Q-1	66	13.5	2345	4.585	10752	15459	26211	-1

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 5-SUBSYSTEM 0

## AIR VEHICLE

	MAN- MON TH S	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL AR S	LABOR + BURDEN \$	ENGR Matl
Q-2 66 Q-3 66							951
TOTAL	13620.5	2340469		10011359	9886093	19897452	916866

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 5-SUBSYSTEM 0

AIR VEHICLE

		MDC	OTHER	SUB		TOTAL
		FFC	CUST	TUTAL	GEA	CUST
Q-1	58	31		259625		259625
0-2	28	221				
Q = 3 Q = 4	50	221		981 099		981099
0 - 1	50	17644	20240	217(2/0		
$\tilde{Q} = 2$	59	T1900	27.507	2110208		31 (0368
$\overline{Q}-3$	59	3389	96374	2342130		22/2120
Q-4	59		/05/11	2342130		2542150
Q-1	60	4203	124088	2075207	39539	2114746
Q-2	60					211110
Q-3	60	308	190853	1497704	28536	1526240
Q-4	60					
Q-1	61	1468	335640	2259323	41985	2301308
Q-2	61					
Q-3	61	507	202928	1891353	35147	1926500
Q-4	61					
Q = 1	62	7441	382729	1688192	28336	1716528
0-2	62	12052	010000			
0-4	62 4 2	13853	313290	1296186	21756	1317942
0 - 1	62	2750	105204	100000	2 . 7	
Q-1 0-2	63	2100	182384	1238035	20700	1258735
$\tilde{\mathbf{n}} = 3$	63	5570	-119201	1200027	20220	1220044
0-4	63		117201	1209031	20229	1230066
0-1	64	16322	154695	2330013	49578	2270501
Q-2	64		23,079	2330013	+ 7 7 1 0	2317371
Q-3	64	27463	153568	356336	7582	363918
Q-4	64					505 510
Q-1	65	4330	145615	470213	7209	477422
Q-2	65					
Q-3	65	735	58246	85679	228£	87965
Q-4	65					
Q-1	66	1	4161	30372	915	31287

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 AIR VEHICLE 5-SUB SYSTEM 0 OTHER SUB TOTAL MPC COST TOTAL GEA COST Q-2 66 Q-3 66 58 1009 30 1039 TOTAL 106624 2267739 23188681 303828 23492509

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 1 AIR VEHICLE

		DESIGN /ENGR HOURS DOLLARS	PROD HOURS DOLLARS	TCOLING AND STE HOURS DOLLARS	TEST /QC HOURS DOLLARS
DESIGN/ENGINEERING		15970542	389215		2012494
LABOR AT \$ 4.921		79719272	2277162	•	QA10000
ENGR BURDEN AT \$	4.463	71224682	2380271		8390825
PRODUCTION		1	6632843		
LABOR AT \$ 3.222			53583877		
SHOP SUPPORT		254520	65703		6961828
LABOR AT \$ 3.036		801624	236401		21068424
PLANN ING			966407		30449
LABOR AT \$ 3.417			3286350		120201
TEST/QC		17505	2058958		608998
LABCR AT \$ '3.531		<b>5919</b> 9	7472494		1949498
MFG BURDEN AT \$	3.918	1101336	79814249		27197817
ENGR MATERIAL		443981			17495087
MFG MATERIAL			37075623		
SUBCONTRACT		61625796	104415066	27555083	896190
MPC		2582197	8627463	1054134	1944631
WIND TUNNEL					4004098
OTHER COST		17320018	657262		1210046
SUB-TOTAL		234878105	299326218	28609217	92695705
GEN & ADMIN		3451754	5852908	506150	1590294
IDWA			23992560		3910650
TUTAL COST		238329859	329671686	29115367	98196649

TTME_PHASED COST			
TTUM-TUMPUN VOOT			
DETAIL - SEE PAGE	II <b>-</b> 62	II-74	II <b>-</b> 86

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#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

#### 4-SYSTEM 1 AIR VEHICLE

	TOTAL
	HOURS
	DOLLARS
DESIGN/ENGINEERING	19373241
LABOR AT \$ 4.921	10070241
ENGR BURDEN AT \$	A 442 01005770
Enon bonden Al S	4.402 01993110
PRODUCTION	16632843
LABOR AT \$ 3.222	53583877
SHOP SUPPORT	7282051
LABOR AT \$ 3.036	22106449
PLANN ING	996856
LABCR AT \$ 3.417	3406551
TEST/QC	2685461
LABOR AT \$ 3.531	9481191
MFG BURDEN AT \$	3.918 108113402
ENGR MATERIAL	17939068
MFG MATERIAL	37075623
SUBCONTRACT	194492135
MPC	14208425
WIND TUNNEL	4004098
OTHER COST	19187326
SUB-TCTAL	656009245
GEN & ADMIN	11401106
IDWA	27903210
•	
TOTAL COST	695313561

TIME-PHASED COST DETAIL - SEE PAGE II-101

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING 4-SYSTEM 1 AIR VEHICLE SUBD OF WORK DESIGN/ENGINEERING

		MAN-	LABOR	LABOR	LABOR	<b>BUR DEN</b>	LABOR +
		MONTHS	HOURS	RATE	DOLLARS	DOLL ARS	BURDEN \$
Q-1	58	732.5	122995	4.618	563005	559607	1127612
Q-2	58						
Q-3	58	3787.0	636217	4.437	2822990	2486700	5309690
Q-4	58						
Q-1	59	5026.5	857878	4.368	3746877	2945404	6692281
Q-2	59						
Q-3	59	8041.5	1415295	4.257	6025412	5084039	11109451
Q-4	59						
0 - 1	60	8908.5	1544068	4.581	7073430	5825267	12898697
Q-2	60						
Q-3	60	8930.0	1500180	4.711	7066939	5547812	12614751
Q-4	60						
Q-1	61	12096.5	2064492	4.786	9880577	7164180	17044757
Q-2	61						
Q-3	61	7986.0	1448090	4.954	7174198	6667669	13841867
Q-4	61						
Q-1	62	7161.0	1222235	5.271	6442288	5619830	12062118
Q-2	62						
Q <b>-3</b>	62	6741.5	1132606	5.269	5967409	5808982	11776391
Q-4	62						
Q-1	63	6195.0	1057254	5.913	6251770	5998781	12250551
Q-2	63						
Q-3	63	6776.5	1138562	5.298	6032522	5835936	11868458
Q-4	63						
Q-1	64	5837.5	996233	5.640	5618539	6191884	11810423
Q-2	64						
Q-3	64	3325.5	585352	5.863	3432171	3871847	7304018
Q-4	64						
Q-1	65	1037.0	179686	6.498	1167512	1177616	2345128
Q-2	65			-			
Q-3	65	365.0	61352	6.682	409939	397016	806955
NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING 4-SYSTEM 1 AIR VEHICLE SUBD CF WORK DESIGN/ENGINEERING

+ \$	LABOR BURDEN	BURDEN Doll Ars	LABOR DOLLARS	LABOR RATE	LABOR HOURS	MAN- MONTHS		
							65	0-4
5 M	803	41 856	38494	4.808	800 <b>7</b>	46.5	66	Q <b>-1</b>
50	600	11000					66	Q-2
56	4	256	200	5.000	40		66	Q-3
54	1509439	71224682	79719272		15970542	92994.0	AL	TOT

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP SUPPORT		
4-SYSTEM	1	AIR	VEHICLE
SUBD OF WORK	DESIGN/ENGINEERI	NG	

#### ON-SITE LABOR

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DGLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$
0-1	58	57.0	9531	3.051	29082	25774	54856
Q-2	58						
Q-3	58	25.5	4198	3.147	13213	12684	2589 <b>7</b>
0-4	58						
Q-1	59		-71	4.295	305	1226	1531
Q-2	59						
10-3	59	246.0	43243	2.926	126508	171670	298178
0-4	59	100.0					
Q-1	60	-120.0	-20814	2.869	-59720	-43877	-103597
Q-2	60	1771 0					
Q-3	<b>6</b> 0	1/1.0	28830	2.886	83199	108807	192006
Q-4 0-1	00 41	101 1	<b>*</b> * * * *				
0-2	01	181.5	30404	2.834	87595	109988	197583
0-2	01 41	11.5	0	-			
0-4	61	40+5	8410	2.852	23986	40033	64019
$\sqrt{-1}$	42	<b>2</b> 0 0	6006				
$\eta = 1$	62	0 <b>0∙</b> 0	5625	3.199	16077	21786	37863
0-2	62	275 5	6 / 73 75 73	2 953			
0-4	62	22202	24131	3.251	178259	220942	399201
0-1	62	251 0	500/1	2 ( ) 2	001000		
0-2	63	301.0	22801	3.412	204272	259107	463379
0-3	63	90.0	15100	2 200	(		
0-4	63	20.00	10108	3.209	48772	99953	148725
$\tilde{0}-1$	64	45 0	7452	2 220	25103	~	
0-2	64	+ <b>J</b> •0	1052	2.220	2548.3	.34555	60038
0-3	64	15 0	2651	2 201	0105	1.000	
0-4	64		2001	3.201	8485	14023	22508
Q-1	65	15.0	2524	2 014	7400	10001	1070-
$\tilde{0}-2$	65		LJL7	2.014	1008	12094	19702
Q-3	65	10.5	1884	3.221	6069	9265	15334

**APRIL 1972** 

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 1 AIR VEHICLE SUBD OF WORK DESIGN/ENGINEERING

# ON-SITE LABOR

LABOR + BURDEN \$	BUR DEN DOLL ARS	LABOR DOLLARS	LABOR RATE	LABOR Hours	MAN- MON THS	
						Q <b>-4</b> 65
5704	3370	2426	3.239	749	4.5	Q-1 66
J170	2210					Q-2 66
-59	-64	5	2.500	2		2-3 66
1902960	1101336	801624		254520	1494.0	TOTAL

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### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC AIR VEHICLE 4-SYSTEM 1 SUBD OF WORK DESIGN/ENGINEERING

# ON-SITE LABOR

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1 Q-2	58 58	1.5	314	3.029	951		951
0-3 0-4	58 58	1.5	158	2.804	443		443
Q-1 Q-2	59 59		-28	2.107	-59		-59
Q-3 Q-4	59 59	13.5	2270	3.163	7181		7181
Q-1 Q-2	60 60	13.5	2363	3.566	8427		8427
Q-3 Q-4	60 60	12.0	2078	3.436	7139		7139
Q-1 0-2	61 61	9.0	1569	3.040	4769		4769
Q-3 Q-4	61 61	3.0	469	3.290	1543		1543
Q-1 Q-2	62 62	1.5	231	4.541	1049		1049
Q-3 Q-4	62 62	7.5	1373	3.137	4307		4307
Q-1 Q-2	63 63	24.0	4009	3.478	13943		13943
Q-3 Q-4	63 63	12.0	1962	3.363	6599		6599
Q = 2 Q = 2	04 64 64	1.5	203	4.030	818		818
Q-4 0-1	04 64 65	1•2	316	4.209	1330		1330
Q = 2 Q = 3	65 65	1 5	40	3.415	139		139
4 2			1.71	2.401	426		456

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

TEST/QC AIR VEHICLE 4-SYSTEM 1 AIR VEHICLE SUBD OF WORK DESIGN/ENGINEERING

LABOR + BURDEN \$	BUR DEN DOLL ARS	LABOR DOLLARS	LABOR RATE -	LABOR HOURS	MAN- MONTHS	
161		161	3.500	46		Q-4 65 Q-1 65 Q-2 66
3		3	3.000	1		Q-3 66
59199		59199		17505	103.5	TOTAL

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYS	STEP	1	1	AIR	VEHICLE
SUBD	OF	WORK	DESIGN/ENGINEERI	NG	

		MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR Matl
Q-1 Q-2	58 58	791.0	132840	4.502	598038	585381	1183419	2315
Q-3 Q-4	58 58	3814.0	640573	4.428	2836646	2499384	5336030	6095
Q-1 Q-2	59 59	5026.5	857779	4.368	3747123	2946630	6693753	7548
Q-3 Q-4	59 59	8301.0	1460808	4.216	6159101	5255709	11414810	24599
Q-1 Q-2	60 60	8802.0	152551 <b>7</b>	4.603	7022137	5781390	12803527	-585
Q-3 Q-4	60 60	9113.0	1531088	4.675	7157277	5656619	12813896	20020
Q-1 Q-2	61 61	12287.0	2096970	4.756	9972941	7274168	17247109	11049
Q-3 Q-4	61 61	8035.5	1456969	4.942	7199727	5707702	13907429	25443
Q-1 Q-2	62 62	7192.5	1227491	5.262	6459414	5641616	12101030	2830
Q-3 Q-4	62 62	7074.5	1188716	5.174	6149975	6029 <b>924</b>	12179899	26544
Q-1 Q-2	63 63	6570.0	1121124	5.771	<b>64</b> 69985	6257888	12727873	18347
Q-3 Q-4	63 63	6878.5	<b>11</b> 55 <b>7</b> 23	5.268	6087893	5935889	12023782	45401
Q-1 Q-2	64 64	5884.0	1004088	5.622	5644840	6226439	11871279	50781
Q-3 Q-4	64 64	3342.0	588319	5.851	3441986	3885870	7327856	116576
Q-1 Q-2	65 65	1052.0	182250	6.449	1175259	1189710	2364969	52183
Q-3 Q-4	65 65	377.0	63367	6.572	416464	406281	822745	13132
Q-1	66	51.0	8802	4.667	41081	45226	86307	17994

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### AIR VEHICLE

4-SYSTEM 1 AL SUBD OF WORK DESIGN/ENGINEERING

	MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR Matl
Q-2 66 Q-3 66		43	4.837	208	192	400	2700
					±72	400	3109
TOTAL	94591.5	16242567		80580095	72326018	152906113	443981

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 AIR VEHICLE SUBD OF WORK DESIGN/ENGINEERING

			TOTAL		OTHER	SUB		
		SUBC	MATERIAL	MPC	COST	TOTAL	G & A	IDWA
0-1 0-2	58 58		2315	125	4005	1189864		
$\overline{Q} - 3$ $\overline{Q} - 4$	58 58	89125	95220	1124	104599	5536973		
Q = 1	59 50	2933925	2941473	78392	475835	10189453		
Q-3	59 50	14910848	14935447	409770	939928	27699955		
Q = 1 Q = 1	50 60	11966051	11965466	709950	2086415	27565368	452518	
Q-3	60 60	7253091	7273111	432970	1535678	22055655	420225	
Q = 1 Q = 1	61 61	5716349	<b>5727</b> 398	164708	2345026	25484241	473574	
Q = 2 Q = 3	61 61	5646678	5672121	163922	1795223	21538695	400251	
Q - 1	62 62	4642746	4645576	147802	2247999	19142407	321304	
Q = 2 Q = 3	62 62	4471422	4497966	144081	1941045	18762991	314934	
Q = 1 Q = 1	63 63	<b>2535</b> 040	2553387	109449	1551769	16942478	283278	
9-2 Q-3	o.∋ 63 ∡ 2	561146	60654 <b>7</b>	22499	-455647	12197181	203937	
Q = 4 Q = 1	63 64	877395	928176	125908	969866	13895229	293746	
Q-3	64 64	12390	128966	46452	893710	8396984	178671	
Q-4 Q-1	64 65 (2	9590	61773	18732	619196	3064670	76427	
Q = 3	65 65		13132	2343	247679	1085899	28972	
Q-4 Q-1	65 65		17994	3733	17692	125726	3787	

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 AIR VEHICLE SUBD OF WORK DESIGN/ENGINEERING

	SUBC	TOTAL MATERIAL	MPC	OTHER COST	SUB TOTAL	GEA	IDWA
Q-2 66	-						
Q-3 66		3709	227		4336	130	
TUTAL	61625796	62069777	2582197	17320018 23	4878105	3451754	

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

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AIR VEHICLE 4-SYSTEM 1 SUBD OF WORK DESIGN/ENGINEERING

		TOTAL Cost
Q-1 Q-2	58 58	1189864
Q-3	58	5536973
Q-1	59 59	10189453
Q-2 Q-3	59 59	27699955
Q-4 Q-1	59 60	28017886
Q-2 Q-3	60 60	22475880
Q-4 Q-1	60 61	25957815
0-2 0-3	61 61	21938946
Q-4 Q-1	61 62	19463711
Q-2 Q-3	62 62	19077925
Q-4 Q-1	62 63	17225756
Q-2 Q-3	63 63	12401118
Q-4 Q-1	63 64	14188975
Q-2 Q-3	64 64	8575655
Q-4 Q-1	64 65	3141097
Q-2 Q-3	65 65	1114871
Q-4 Q-1	65 66	129513

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 AIR VEHICLE SUBD OF WORK DESIGN/ENGINEERING

> TOTAL COST

0-2 66

6 1

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Q-3 66 4466

TOTAL 238329859

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

DESIGN/ENGINEERING 4-SYSTEM 1 SUBD OF WORK PRODUCTION AIR VEHICLE

		MAN- MON THS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58		8	4.625	37	36	73
Q-2	58						
4-3	58	81.0	13518	5.414	73189	9721	82910
Q - 4	58						
Q-1	59	97.5	16629	5.456	90721	14539	105260
Q = 2	59						
Q-3	59	174.0	30569	5.273	161199	34800	195999
Q-4	59						
Q-1	60	67.5	11814	5.500	64976	16616	81592
0-2	60						
Q-3	60	1.5	234	4.060	950	-759	191
Q-4	60						• · · •
Q-1	61	3.0	608	4.942	3005	3088	6093
Q-2	61						
Q-3	61	4.5	725	3.607	2615	960	3575
Q-4	61						0.10
Q-1	62	12.0	2131	4.363	9298	11 224	20522
Q-2	62				220	1124.1	
Q-3	62	51.0	8606	4.356	37497	42640	00166
0 - 4	62				1041	72000	01222
0 - 1	63	55.5	9578	4 720	45204	50.000	0/ 01 0
$\bar{0}-2$	63	· · · · · · · · · · · · · · · · · · ·	7710	7012.3	47290	20.922	90218
0-3	63	-363-0	-61028	5 405	-325320	- 12040	2/0107
0-4	63	505.0	01020	J • 47 J	-222228	-12000	-349197
$\tilde{0} = 1$	64	97 5	16604	1. 604	70000	100357	13034
0-2	64		10034	7.074	10005	100357	1/8/16
0-3	64	1345 5	224702	5 000	12/0//1	1/01000	<b>AAAAAAAAAAAAA</b>
0-4	64	134343	200195	2.239	1240561	1481893	2122554
$\tilde{\mathbf{n}}_{-1}$	65	227 F	FOLDE	6 124	22222		
N	45	22102	28423	5.136	300082	376259	676355
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	00 4 G	3(1.0	(2010				
Q-3	60	201.0	43812	11.510	504257	251470	755727

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CUNTRACT NAS9-12100

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# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

DESIGN/ENGINEERING 4-SYSTEM 1 SUBD OF WORK PRODUCTION AIR VEHICLE

LABOR + BURDEN \$	BUR DEN DOLL ARS	LABOR DOLLARS	LABOR RATE	LABOR HOURS	MAN- MONTHS	
690	345	345	3•485	ġð		Q-4 65 Q-1 66 Q-2 66 Q-3 66
4657433	2380 271	2277162		389215	2226.0	TOTAL

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PRODUCTION 4-SYSTEM 1 AIR SUBD OF WORK PRODUCTION

#### AIR VEHICLE

		MAN-	LABOR	LABOR	LABOR	<b>BUR DEN</b>	LABOR +
		MONTHS	HOURS	RATE	DOLLARS	DOLL ARS	BURDEN \$
Q-1	60	45.0	7756	4.702	36469	15299	51768
Q-2	60						
0-3 0-4	60 60	76.5	12911	2.650	34218	23865	58083
Q-1	61	3259.5	556180	3.143	1747956	2094250	3842206
Q-2	61						
Q-3	61 61	10192.5	1848287	3.152	5825301	7400880	13226181
0-1	62	15639.0	2669161	3.075	8207356	11012512	19219868
Q-2	62						
Q-3 0-4	6 <u>2</u> 62	13857.0	2328074	3.202	7455133	10492911	17948044
Q-1	63	12666.0	2161579	3.286	7103831	9745299	16849130
0-2	63						
Q - 3 Q - 4	63 63	14742.0	2476741	2.826	6998049	11455827	18453876
0-1	64	14718.0	2511784	3.580	8992441	14789108	23781549
Q-2	64						
Q-3 Q-4	64 64	6741.0	1186480	3.365	3992935	6429221	10422156
Q-1	65	4222.5	731845	3.509	2568369	3860 465	6428834
Q-2	65						
Q-3	65	846.0	142045	4.378	621819	837069	1458888
Q-4	65						
Q-1	66					10096	10096
Q-2	66						
Q-3	66					7	7
тот	AL	97005.0	16632843		53583877	78166809	131750686

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP SUPPORT		
4-SYSTEM	1	AIR	VEHICLE
SUBD OF WORK	PRODUCTION		•

	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DCLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q-3 64	261.0	45842	3.628	166336	233354	399690
Q-1 65 0-2 65	114.0	19861	3.528	70065	95548	165613
Q-3 65					-819	-819
TOTAL	375.0	65703		236401	328083	564484

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

# PLANNING

4-SYSTEM 1 SUBD OF WORK PRODUCTION

#### AIR VEHICLE

		MAN- MONTHS	LABOR			BUR DEN	LABOR +
			1.007.0	NALL	DULLANS	DULLAKS	DURDEN D
Q-3	58	12.0	1905	2.998	5711		5711
ର–4	58						2111
Q-1	59	55.5	9569	2.960	28324		28324
Q-2	59						2002 (
Q-3	59	133.5	23401	2.989	69940		69940
Q-4	59						
Q-1	60	270.0	46837	3.160	147990	483	148473
Q-2	60						
Q-3	60	544.5	91472	3.050	278949		278949
Q-4	60						
0-1	61	969.0	165424	3.022	499849	74882	574731
Q-2	61						
Q-3	61	876.0	158836	2.917	463256	78426	541682
Q-4	61						
Q-1	62	828.0	141246	2.980	420899	79970	500869
Q-2	62						
Q-3	62	751.5	126222	2.977	375747	86277	462024
Q-4	62						
Q-1	63		45	3.800	171	207	378
Q - 2	63						
Q-3	63	519.0	87095	7.303	636031	438506	1074537
Q = 4	5.5						
	04	472.5	80585	3.096	249479	394254	643733
	64	104.0					
€-3 0 4	64	126.0	22279	3.325	74073	<b>111</b> 685 <sup>.</sup>	185758
Q-4 0 1	04 (5	<b>E1</b> 0					
Q-1 0-2	02	<b>91</b> •0	8912	3.120	27804	42183	69987
Q-2	00 4 E	15.0	2504				
6-5 0-6	0) 4 E	15.0	2586	3.120	8069	12468	20537
Q-4 0-1	00		-		<b>.</b>		
9–1	00		- /	8.285	58	16	74
тот	AL	5623.5	966407		3286350	1319357	4605707

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

AIR VEHICLE

			TEST/QC	
4-SY \$	STEN	1	1	
SUBD	OF	WORK	PRODUCTION	

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		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q <b>-3</b>	58	6.0	931	4.908	4569		4569
Q-4	58						,
Q-1	59	6.0	898	4.257	3823		3823
Q-2	59						
Q-3	59	25.5	4471	4.665	20855		20855
Q-4	59						
Q-1	<b>6</b> 0	51.0	8936	5.052	45145		45145
<b>Q-</b> 2	60						
Q-3	<b>6</b> 0	9.0	1591	3.901	6207		6207
Q-4	60						
Q-1	61	232.5	39568	3.133	123963		123963
Q-2	61						
Q-3	61	739.5	133975	3.147	421666		421666
Q-4	61						
Q-1	62	1521.0	259621	3.230	83864C		838640
Q-2	62						
Q-3	62	1746.0	293411	3.424	1004639		1004639
Q-4	62	<b>.</b>					
Q-1	6.3	1623.0	276938	3.625	1003816		1003816
0-2	63						
Q-3	63	1996.5	335427	4.273	1433282		1433282
0-4	63	224					
0-1	64	2269.5	387399	3.589	1390543		1390543
<b>₩</b> −2	64	1124 0					
Q-5	04	1134.0	199504	3.598	717802		717802
Q-4	64 7 E	F( 2 0					
	65		94048	3.850	362050		362050
0 2	0) (E	170 0					
8-3	כס	132.0	22240	4.294	95494		95494
тот	AL	12034.5	2058958		7472494		7472494

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 SUBD OF WORK PRODUCTION

AIR VEHICLE

		MAN-	LABOR	LABOR	LABOR	<b>BUR DEN</b>	LABOR +	ENGR
		MONTHS	HOURS	RATE	DOLLARS	DOLLARS	BURDEN \$	MATL
Q-1	58		8	4.625	37	36	73	
Q-2	58						•	
Q - 3	58 59	99+0	16354	5.104	83469	9721	93190	
0 1	20	150 0						
Q = 1 Q = 2	59 59	159.0	27096	4.535	122868	14539	137407	
0-3	50	333 0	59441	6 212	261001	~ / ^ ~		
0 - 4	59	22200	20441	4.012	201994	34800	286794	
0-1	60	433.5	75343	3.010	204500	22200	224070	
Q-2	60	· · · · · · · · · · · · · · · · · · ·	12010	3. 110	294000	36370	320710	
0-3	60	631.5	106208	3.016	326324	23106	262620	
Q-4	60			21010		CO 100	040400	
Q-1	61	4464.0	761780	3.117	2374773	2172220	4546993	
Q-2	61							
Q-3	61	11812.5	2141823	3.134	6712838	7480266	14193104	
Q-4	61							
Q-1	62	18000.0	3072159	3.085	9476193	11103706	20579899	
Q-2	62							
Q-3	62	16405.5	2756313	3.219	8873006	10621856	19494862	
0 - 4	0Z 20	1/2// 5						
0-2	00 40	14344.5	2448140	3.330	8153114	9796428	17949542	
Q = 2	0.0 4 7	14004 5	2020225	~ ~ ~ ~ ~		<b>.</b>		
Q = 3 Q = 4	62	10094.0	2838235	3.077	8732033	11880465	20612498	
$\tilde{0}_{-1}$	54	17557 5	2004442	2 576	10710000	1 5000 710		
0 - 2	64		2330402	3+214	10/10822	15283719	25994541	
0-3	64	9607.5	1600909	3 443	(101007	005/150	14447040	
$\tilde{Q}-4$	64		10,00,0	0.002	0191007	0200103	14447960	
0-1	65	5268.0	913091	3.645	3328384	4374455	7702920	
Q-2	65		- <u>.</u>		<i></i>	TUTTU	1102039	
Q-3	65	1254.0	210683	5.836	1229639	1100188	2329827	
Q-4	65		· · · · · · ·				LJC/UC1	
Q-1	66		92	4.380	403	10457	10860	

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

AIR VEHICLE

4-SYSTEM 1 SUBD OF WORK PRODUCTION

	MAN- MONTHS	LABOR Hours	LABOR RATE	LABOR Dollars	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
Q-2 66 Q-3 66					7	7	
TOTAL	117264.0	20113126		66856284	82194520	149050804	

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 SUBD OF WORK PRODUCTION

AIR VEHICLE

	•	MFG		TOTAL		OTHER	SUB	
		MATL	SUBC	MATERIAL	MPC	COST	TOTAL	G&A
Q = 1	58 58					5641	5714	
Q-3	58 58		29946	29946	265	112960	236361	
Q-4 Q-1	58 59		555 <b>137</b>	555 <b>1</b> 37	14717	190600	897861	
Q-2 Q-3	59 59		6449252	6449252	176221	269832	7182000	
0-4	59			0117272	110221	207032	1102077	
Q-1 Q-2	60 60	400	7579016	7579416	449708	5714	8361816	159640
Q - 3	60 60	15609	7951934	<b>7</b> 967543	473857	19886	8804716	181734
Q - 1	61	1001899	12070287	13072186	430143	26 28 3	18075605	394142
Q-2 Q-3	61	4707321	15525866	20233187	842176	12977	35281444	734029
Q-4 Q-1	62	4903221	18404312	23307533	971161		44858593	800299
Q=2 Q=3	62 62	5119756	13066692	18186448	818198		38499508	675884
Q - 1 Q - 1	63 63	5800375	10822277	16622652	1030995		35603189	643803
Q-2 Q-3	63 63	<b>655</b> 906 <b>7</b>	5678913	12237980	828725	2424	33681627	610325
Q-1 Q-2	64 64	5706444	6069327	11775771	1441 856	9882	39222050	923323
Q-3	64	2336960	170650	2507610	905921	1063	17862554	413240
Q-1	64 65	508486	41457	549943	165626		8418408	237946
Q-3	65 65	440881		440 88 1	78656		2849364	78986
Q-4 Q-1	65 66	5227		5227	1084		17171	517

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 SUBD OF WORK PRODUCTION

	MFG Matl	SUBC	TOTAL MATERIAL	MPC	OT HER COST	SUB Total	G&A
Q-2 66							
Q-3 66	-30023		-30023	-1846		-31862	-960
TOTAL	37075623	104415066	141490689	8627463	657262	299826218	5852908

AIR VEHICLE

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

4-SYSTEM 1 SUBD OF WORK PRODUCTION

AIR VEHICLE

		IDWA	TOTAL COST
0-1	58		5714
Q-2	58		
Q-3	58		236361
Q-4	58		
Q-1	59	11147	909008
Q-2	59		_
Q-3	59	4243	7186342
0-4	59	(100	0.000.007
0-2	00	6400	8527856
0-2	60 60	50151	00/5/01
0-4	60	59151	9042601
Q-1	61	2512358	20982105
Q-2	61		
Q-3	61	3923564	39939037
Q-4	61		
Q-1	62	2924005	4858289 <b>7</b>
Q-2	62		
Q-3	62	2610066	41785458
Q-4	62		
Q-1	63	2496459	38743451
Q-2	63		
Q-3	63	3068199	37360151
Q-4	63		
	64	4803875	44949248
0-2	04	15/1/17	10007/11
0 1	04 47	1201011	1983/411
∾=+ ∩=1	65	6 <b>7</b> 7 7	0661001
w=1 ∩2	65	4121	0001001
0-3	65	6749	2025000
Q-4	65	0147	2222633
0-1	66		17688

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

> > AIR VEHICLE

4-SYSTEM 1 SUBD OF WORK PRODUCTION

	TOTAL
IDWA	COST

Q-2 66

- Q-3 66 -32822
- TOTAL 23992560 329671686

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

AIR VEHICLE

4-SYSTEM 1 SUBD OF WORK TOOLING AND STE

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DCLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	SUBC
Q-1 59							40154
Q-2 59							42120
Q-3 59							708102
Q-4 59							100172
Q-1 60							2645053
Q-2 60	•						2010000
$\sqrt{-3}$ 60							2438699
Q = 4 60							
Q = 1 01 Q = 2 41							4959380
0-3 61					×.		
0-4 61							8019599
0 - 1 62							
Q = 2 - 62							2889461
0 - 3 62							
9-4 62							3760321
Q-1 63							
Q-2 63							1008179
Q-3 63							553004
Q-4 63							221934
Q-1 64							663663
Q-2 64							441443
Q-3 64							67059
Q-4 64							01057
Q-1 65							17607
TOTAL				••			0755500-
							21222083

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 AIR VEHICLE SUBD OF WORK TOOLING AND STE

		MPC	SUB Total	GεA	TOTAL COST
Q-1 9-2	59 59	1117	43273		43273
Q-3 Q-4	59 59	19349	727541		727541
0-1 0-2	60 60	156926	2801979	60236	2862215
Q-3 Q-4	60 60	144690	2583389	89969	<b>267</b> 3358
Q-1 Q-2	61 61	142085	5101465	97866	5199331
Q-3 0-4	61 61	229763	8249362	130732	8380094
Q-1 Q-2	62 62	91829	2981290	40325	3021619
$\overline{Q} - \overline{3}$ $\overline{Q} - 4$	62 62	119397	3879718	47418	3927136
$\hat{Q} = 1$ $\hat{Q} = 2$	63 63	42806	1050985	17362	1068347
Q-3 Q-4	63 63	17924	5 <b>7</b> 5858	9140	584998
Q = 1 Q = 2	64 64	60623	502066	10583	512649
Q-3 Q-4	64 64	21881	88940	1892	90832
Q-1	65	5744	23351	623	23974
тот	AL	1054134	28609217	506150	29115367

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# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESIGN/ENGI	NEERING
4-SYSTEM	1	
SUBD DF WORK	TEST/QC	AIR VEHICLE

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DCLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	<b>5</b> 8	48.0	8007	4.442	35570	36430	72000
Q-2	58						
Q-3	58	190.5	32064	4.157	133284	125443	258727
Q-4	58						
Q-1	59	286.5	48836	4.175	203866	167509	371375
Q-2	59						
Q-3	59	213.0	3 <b>7</b> 359	4.110	153546	134425	287971
Q-4	59						
Q-1	60	456.0	78933	4.496	354871	276037	630908
Q-2	60						
Q-3	60	1480.5	248723	4.057	1009154	924883	1934037
Q-4	60						
Q-1	61	1807.5	308500	3.707	1143641	984397	2128038
0-2	61						
Q-3	61	1563.0	283438	4.311	1221925	1280733	2502658
Q-4	61						
Q-1	62	2009.5	343015	4.266	1463370	1479238	2942608
Q-2	62						
Q-3	62	1760.5	29579 <b>7</b>	4.332	1281486	1308685	2590171
Q-4	62						
Q-1	63	592.0	100967	4.802	484823	563253	1048076
Q-2	63					,	
Q-3	63	475.0	79859	4.690	374574	401289	775863
Q-4	63			••			
Q-1	64	341.5	58294	4.785	278963	353627	632590
Q-2	64						
Q-3	64	176.5	31037	4.350	135019	183541	318560
Q-4	64						
Q-1	65	238.5	41322	2.278	94136	107269	201405
Q-2	65						
Q-3	65	112.5	18933	2.995	56703	68681	125384

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

DESIGN/ENGINEERING 4-SYSTEM 1 SUBD OF WORK TEST/QC AIR VEHICLE

	and new	LABOR	I ABOR	LABOR	MAN-		
BURDEN \$	DOLLARS	DOLLARS	HOURS RATE D	HOURS	MONTHS		
						4 65	Q-4
-8404	-3488	-4916	3.847	-1278	-7.5	1 66	Q-1
•						2 66	0-2
-2254	-1127	-1127	3.500	-322	-1.5	3 66	Q-3
16809713	8390825	8418888		2013484	11742.0	OTAL	TOT

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP SUPPORT		
4-SYSTEM	1		
SUBD OF WORK	TEST/QC	AIR	VEHICLE

		MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1	58		21	2.476	52	73	125
Q-2	58						
Q-3	58	393.0	65963	2.959	195215	223717	418932
Q-4	58					~~~~	120302
Q-1	59	2580.0	440392	2.959	1303166	1570961	2874127
Q-2	59						
Q-3	59	2632.5	463330	2.922	1353656	1812754	3166410
Q-4	59						2100410
Q-1	60	3711.5	643285	2.953	1905775	2335373	4241148
Q-2	60					8. 6 hr 8. 9 hr 10	7271170
ର– 3	60	6366.0	1069448	3.036	3246734	3914402	7161126
Q-4	60						1101130
Q-1	61	11555.5	1972145	3.063	6041034	7307719	12248752
Q-2	61					1301112	13340123
Q-3	61	5615.5	1018267	3.042	3098001	4515108	7613100
Q-4	61					1212200	1015103
Q-1	62	3593.5	613359	2.885	1769570	2199679	3969269
Q-2	62				2100010	14 A. J. M. 9 J.	3707249
Q-3	62	1952.5	328056	3.091	1013895	1375198	2280002
Q-4	62					10 ( ) 1 ) (	2303035
Q-1	63	764.5	130411	3-201	417503	554026	071520
Q-2	63				121303	2,24020	211763
Q-3	63	746.0	125339	3.446	431945	890.005	1321050
Q-4	63				,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	0,0000	1721970
Q-1	64	511.5	87277	2.842	248065	384779	632844
0-2	64		,	200.2	210000	301113	032077
Q-3	64	100.5	17699	.706	12500	40605	52105
Q-4	64						13101
Q-1	65	-72.5	-12525	. 844	10586	25570	36156
Q-2	65	. –				<i>L &gt; &gt;</i> 10	0100
Q-3	65		74	291.284	21555	37307	58862

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 1 SUBD OF WORK TEST/QC AIR VEHICLE

LABOR + BURDEN \$	BURDEN DOLLARS	LABOR DOLLARS	LABOR RATE	LABOR HOURS	MAN- MONTHS		
						65	Q-4
-697	98	-795	1.128	-705	-4.5	66	Q-1
						66	Q-2
20	53	-33	4.125	-8		66	Q-3
48255851	27187427	21068424		6961828	40445.5	TAL	тот

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### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# PLANNING

4-SYSTEM 1 SUBD OF WORK TEST/QC

#### AIR VEHICLE

	MAN- MON TH S	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1 63		-82	3.317	-272	-141	-413
Q = 2 - 63 Q = 3 - 63	160.0	26848	3.970	106582	2586	109168
Q-4 63 Q-1 64	8.5	1451	3.747	5437	3399	8836
Q-2 54 Q-3 64	8.5	1449	3.738	5417	3410	8827
Q-4 64 Q-1 65	3.0	548	3.880	2126	791	2917
Q-2 65 Q-3 65	1.5	219	3.881	85 0	316	1166
Q-4 65 Q-1 66		16	3.813	61	23	84
TOTAL	181.5	30449		120201	10384	130585

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

AIR VEHICLE

	TEST/QC
4-SYSTEM	1
SUBD OF WORK	TEST/OC

1

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1	58						
Q-2	58						
Q-3	53	9.0	1565	3.539	5538		5538
Q-4	58						
Q-1	59	45.0	7788	3.005	23404		23404
Q-2	59						
Q-3	59	114.0	20038	3.104	62191		62191
Q-4	59						
Q-1	60	242.5	42036	2.970	124840		124840
Q-2	60						
Q-3	60	661.5	111179	3.217	357703		357703
Q-4	60						
Q-1	61	1335.0	227917	3.248	740251		740251
Q-2	61						
Q <b>-3</b>	61	652.0	118171	3.140	371109		371109
Q-4	61						
Q-1	62	193.5	33089	3.229	106859		106859
Q-2	62						
Q-3	62	115.5	19411	3.091	59996		59996
Q-4	65						
Q-1	63	38.5	6518	3.459	22547		22547
Q-2	63						
Q-3	63	46.0	7651	3.377	25838		25838
Q-4	63						
Q-1	64	52.0	8866	3.712	32913		32913
Q-2	64		,				
Q-3	64	2.5	451	• 54 5	246		246
Q-4	64						
Q-1	65	21.0	3649	3.782	13802		13802
Q-2	65					,	
Q-3	65	4.0	620	3.313	2054		2054

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 1 AIR VEHICLE SUBD OF WORK TEST/QC

LABOR + BURDEN \$	BUR DEN DOLL ARS	LABOR DOLLAR S	LABOR RATE	LABOR HOURS	MAN- MONTHS	
						Q-4 65
121	-40	161	4.237	38		Q-1 66
and the star						Q-2 66
92	46	46	4.182	11		Q-3 66
1949504	6	1949498		6C8998	3532.0	TOTAL

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 SUBD OF WORK TEST/QC AIR VEHICLE

		MAN-	LABOR	LABOR	LABOR	BURDEN	LABOR +	ENGR
		MONTHS	HOUR S	RATE	DOLLARS	DOLL ARS	BURDEN \$	MATL
Q-1	58	48.0	8028	4.437	35622	36503	72125	32
<b>Q-</b> 2	58							_
Q-3	58	592.5	\$9592	3.354	334037	349160	683197	121438
0-4	58							
Q = 1 Q = 2	59 59	2911.5	497016	3.079	1530436	1738470	3268906	371316
0-3	59	2959.5	520727	2 014	1640202	1047170	251(572	200020
Q-4	59		JEOILI	<b>J</b> + 1 J + C I +	1 20 2 2 2 2	124/1/2	2210215	202932
0-1	60	4410-0	764254	3, 121	2285486	2411410	4004 904	1443317
0-2	60	111000	101274	3*161	2000400	2011410	OCODER	1401310
0-3	60	8508-0	1429350	3, 228	4613501	4930 395	0452974	1040525
0-4	60	0,000000	3. 8 <i>4. 2 4 2</i> 4	- <b></b>	1010001	+057205	9422010	1046555
0 - 1	61	14698.0	2502562	2 150	707/074	0202114	1/0170/0	37//005
0 - 2	61	1107080	200002	J • 1 3 3	1924920	0292110	1021/042	2140905
0-3	61	7830.5	1419876	3.304	4691035	5795841	10496976	2144007
0-4	61		X I Y OF U	3.304	40/1000	JUJUTI	10400010	J144001
Q-1	62	5795-5	989463	3, 375	2320700	3678617	7019716	1110220
0 - 2	62	242042	202103			2010311	1010110	1119529
N-3	62	3828.5	643264	3-662	2355377	2683883	5030260	1482225
$\hat{\mathbf{Q}}$ -4	62		0.0201	JUGGE	11000	20000000	<b>JU</b> J7 <b>Z</b> UU	1403233
0-1	63	1395.0	237814	3,888	924601	1117138	2041739	1416426
$\hat{\mathbf{Q}} = 2$	63				JE GOL	111,130	2041133	1410420
$\tilde{c}-\tilde{3}$	63	1427.0	239697	3,917	938939	1293880	2232819	2622128
Q-4	63				م میں در میں میں میں	12,10,000		2022120
Q-1	64	913.5	155888	3.627	565378	741 805	1307183	456499
Q-2	64							120199
Q - 3	64	288.0	50636	3.025	153182	227556	380738	379313
0-4	64		,					
Q-1	65	190.0	32994	3.657	120650	133630	254280	564041
0-2	65	×						
Q-3	65	118.0	19846	4.090	81162	106304	187466	213619
Q-4	65					-		
Q-1	66	-12.0	-1929	2.846	-5489	-3407	-8896	36894

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SY Sube	STEM 1 DOF WORK T	EST/QC	AIR V	EHICLE				
	MAN- MONTHS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$	ENGR MATL	
Q-2 66 Q-3 66 Q-4 66 Q-1 67 Q-2 67 Q-3 67	-1.5	-319	3.492	-1114	-1028	-2142	26242	
TOTAL	55901.0	9614759		31557011	35588642	67145653	17495087	

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

AIR VEHICLE

4-SYSTEM 1 SUBD OF WORK TEST/QC

		SUBC	TOTAL MATERIAL	MPC	W IND TUNNEL	OT HER COST	TOTAL D/C \$	SUB Total
Q - 1	58 59		32	2	170368	1293	171661	<b>243</b> 820
0-2	58		101400	(417	500500	0440		
a - 4	58		121430	0042	283228	-9669	573859	1385136
$\tilde{Q}-1$	59		371316	31440	610704	76663	607267	4350030
Q-2	59			02110	010101	10005	101101	4339029
Q-3	59	9303	292235	24211	695957	58795	754752	4587770
Q-4	59							
Q-1	60	166202	1627518	202067	338115	77173	415288	7241769
Q-2	60	1(2250	1010505					
0-3 0-4	00 40	162050	1210585	147513	403 08 9	62100	465189	11276163
0 - 1	61	220498	2967602	220200	204523	12001	200/11/	107001/0
$\tilde{Q} = 2$	61	220470	2001400	200009	290020	12891	309414	19732168
Q-3	61	63814	3208701	267451	237221	58753	295974	14259002
Q-1	62	32400	1151729	89328	<b>2</b> 0 <b>7</b> 559	74575	282134	8541907
Q-2	62							
Q-3	62	26054	1509289	118457	177914	359688	537602	7204608
0-1	62	72413	1400000	1/0/57	140054			
0-2	63	12412	1402022	142 55 /	148254	169674	317928	3992163
0-3	63	38968	2661.096	259588	112797	311104	4 20001	5502404
Q-4	63		2001070	223200	LUIT	211124	427371	JJ03494
Q-1	64	103488	559 <b>9</b> 8 <b>7</b>	62872	- 1	-28401	-28400	<b>1901</b> 642
Q-2	64							
Q-3	64		379313	138000	-1	-23441	-23442	874609
Q-4	64 45		E(10(1)	1/0717				
u-1 0-2	00 65		204041	108/1/	1	6129	6130	993168
a 2 D-7	65		213619	38114	•	2452	2452	143450
0-4	65		CTOCT)	JULLY		2423	2423	441002
0-1	66		36894	7651	11740	176	11916	47565

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	4-SYSTEM 1 SUBD OF WORK TEST/QC			AIR V	VEHICLE			
		SUBC	TGTAL MATERIAL	мрс	WIND TUNNEL	OT HER Cost	TOTAL 0/C \$	SUB Total
Q-2 Q-3 Q-4 Q-1 Q-2 Q-3	66 66 67 67 67		26242	1612	4328		4328	30040
TO	<b>FAL</b>	<b>8961</b> 90	18391277	1944631	4004098	1210046	5214144	92695705
#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

AIR VEHICLE

4-SYSTEM 1 SUBD OF WORK TEST/QC

	БАЗ	IDWA	COST
Q = 1 58 Q = 2 58			243820
Q-3 58 Q-4 58			1385136
<b>2−1</b> 59 <b>2−2</b> 59		12067	4371096
2-3 59 2-4 59		4605	4592375
Q-1 60 Q-2 60	143158	6935	7391862
Q-3 60 Q-4 60	220716	37147	11534026
)-1 61 )-2 61	400975	1860333	21993476
Q-3 61 Q-4 61	301557	1989563	16550122
2-1 62 2-2 62	143261		8685168
Q-3 62 Q-4 62	120837		7325445
<b>⊋−1 63</b> <b>⊋−2 63</b>	66749		4058912
2-3 63 2-4 63	93355		5676849
Q-1 64 Q-2 64	40462		1942104
Q-3 64 Q-4 64	18609	,	893218
Q-1 65 Q-2 65	26497		1019665
Q-3 65 Q-4 65	11782		453434
2-1 66	1431		48996

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# APRIL 1972

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM	1	AIR	VEHICLE
SUBD OF WORK	TE ST/QC		

		GεA	IDWA	TOTAL COST
Q-2	66			
2-3	66	905		30945
2-4	66			
2-1	67			
ຊ – 2	67			
Q-3	67			

TOTAL 1590294 3910650 98196649

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## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM

#### AIR VEHICLE

ON-SITE LABOR

DESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-1 Q-2	58 58	780.0	131010	4.607	603612	596073	1199685
Q-3 Q-4	58 58	4058.0	681799	4.443	3029463	2621864	5651327
Q-1 Q-2	59 59	5410.5	923343	4.377	4041464	3127452	7168916
Q-3 Q-4	59 59	8427.5	1483223	4.275	6340157	5253264	11593421
Q-1 Q-2	60 60	9432.0	1634815	4.584	7493277	6117920	13611197
Q-3 Q-4	60 60	10411.5	1749137	4.618	8077043	6471936	14548979
Q-1 Q-2	61 61	13908.0	2373600	4.646	11027223	8151665	19178888
Q-3 Q-4	61 51	9553.C	1732253	4 • 84 8	8398738	7949362	16348100
Q-1 Q-2	62 62	9183.5	1567381	5.050	7914956	7110292	15025248
Q-3 Q-4	62 62	8553.5	1437009	5.071	7286382	7160335	14446717
Q-1 Q-2	63 63	6842.5	1167799	5.807	6781889	6612956	13394845
Q-3 Q-4	63 63	6889.0	1157393	5.246	6071767	6223357	12295124
Q-1 Q-2	64 64	6277.0	1071221	5.579	5975861	6645868	12621729
Q-3 Q-4	64 64	4847.5	853182	5.635	4807851	5537281	10345132
Q-1 Q-2	65 65	1612.5	279433	5.589	1561744	1661144	3222888
Q-3 Q-4	65 65	738.5	124097	7.824	970899	717167	1688066

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	4-SYSTEM		DESIGN/ENGINEERING						
		JILIT	AIR VEHICLE						
			ON-SITE LAB	OR					
		MAN- MONTHS	+ LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$		
Q-1 Q-2	66 66	39.0	6828	4.968	33923	38713	72636		
Q-3	66	-1.5	-282	3.287	-927	-871	-1798		
TOT	AL	106962.0	18373241		90415322	81995778	172411100		

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM

1

AIR VEHICLE

ON-SITE LABOR

PRODUCTION

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1	60	45.0	7756	4.702	36469	15299	51768
Q-2	60						
Q-3	60	76.5	12911	2.650	34218	23865	58083
Q-4	60						
0 - 1	61	3259.5	556180	3.143	1747956	2094250	3842206
Q-2	61						
Q-3	61	10192.5	1848287	3.152	5825301	7400880	13226181
Q-4	61	16/20 0		0.075	000705/		100100/0
Q - I	62	15639.0	2669161	3.015	8207355	11012512	19219868
Q-2	62	12057 0	2222074	2 202	7/55122	10402011	17040044
Q-0	62 40	12821.0	2328014	3 • 202	1400100	10492911	11940044
y = 4 0 = 1	62 43	12666 0	2141570	3 286	7103931	0745700	16869130
0-2	63	12000.0	2101010	2.200	110,2051	JT 4 J 2 J 2	10347130
0-3	63	14742.0	2476741	2.826	6998049	11455827	18453876
0-4	63	1					
0 - 1	64	14718.0	2511784	3.580	8992441	14789108	23781549
Q-2	64			,			
Q-3	64	6741.0	1186480	3.365	3992935	6429221	10422156
Q-4	64				,		
Q-1	65	4222.5	731845	3.509	2568.369	3860465	6428834
Q-2	65						
Q-3	65	846.0	142045	4.378	621819	837069	1458888
Q-4	65						
Q-1	66					10096	10096
Q-2	66					_	_
Q-3	66					7	7
TO	[ AL	97005.0	16632843		53583877	78166809	131750686

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM

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SHOP SUPPORT 1

AIR VEHICLE

#### ON-SITE LABOR

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DGLLARS	BUR DEN DOLL AR S	LABOR + BURDEN \$
Q-1 Q-2	58 58	57.0	9552	3.050	29134	25847	54981
Q-3 Q-4	58 58	417.0	70161	2.971	208428	236401	444829
ହ−1 ହ−2	59 59	2580.0	440321	2.960	1303471	1572187	2875658
Q-3 Q-4	59 59	2878.0	506573	2.922	1480164	1984424	3464588
Q-1 Q-2	60 60	3591.5	622471	2.966	1846055	2291496	4137551
Q-3 Q-4	60 60	6537.0	1098278	3.032	3329933	4023209	7353142
Q-1 Q-2	61 61	11737.0	2003054	3.060	6128629	7417707	13546336
Q-3 Q-4	61 61	5662.0	1026677	3.041	3121987	4555141	7677128
Q-1 Q-2	62 62	3623.5	618384	2.888	1785647	2221465	4007112
Q-3 Q-4	62 62	2278.0	382793	3.114	1192154	1596140	2788294
Q-1 Q-2	63 63	1115.5	190272	3.268	621775	813133	1434908
Q-3 Q-4	63 63	836.5	140538	3.421	480717	989958	1470675
Q-1 Q-2	64 64	556.C	94929	2.882	273548	419334	692882
Q-3 Q-4	64 64	376.0	66192	2.830	187321	287982	475303
Q-1 Q-2	65 65	57.0	9860	8.951	88259	133212	221471
Q-3 Q-4	65 65	12.0	1958	14.108	27624	45753	73377

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## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM		EM	SHOP 1	SUPPOR	T	VEHICLE		
			ON-	SITE LA	BOR		. ,	
		MAN- MON TH S		LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q-1 Q-2	66 66			44	37.068	1631	3468	5099
Q-3	66			-6	4.667	-28	-11	-39
тот	TAL 4	+2314.0	7	282051		22106449	28616846	50723295

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYS TEM

PLANNING 1

AIR VEHICLE

## ON-SITE LABOR

		MAN- MON TH S	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 Q-4	58 58	12.0	1905	2.998	5711		5711
Q-1 Q-2	59 59	55.5	9569	2.960	28324		28324
Q-3 Q-4	59 59	133.5	23401	2.989	69940		69940
Q-1 Q-2	60 60	270.0	46837	3.160	147990	483	148473
Q-3 Q-4	60 60	544.5	91472	3.050	278949		278949
Q-1 Q-2	61 61	969.0	165424	3.022	499849	74882	574731
Q-3 Q-4	61 61	876.0	158836	2.917	463256	78426	541682
Q - 1 Q - 2	62 62	828.0	141246	2.98C	420899	79970	500869
Q-3 Q-4	62 62	751.5	126222	2.977	375747	86277	462024
Q-1 Q-2	63 63		-37	2.730	-101	66	-35
Q-3 Q-4	63 63	678.0	113943	6.517	742613	441092	1183705
Q-1 Q-2	64 64	481.0	82036	3.107	254916	397653	652569
Q-3 Q-4	64 64	135.0	23728	3.350	79490	115095	194585
Q-1 0-2	65 65	55.0	9460	3.164	29930	42974	72904
Q-3	65 65	16.5	2805	3.180	8919	12784	21703
Q-1	66		9	13.222	119	39	158
тот	AL	5805.5	996856		3406551	1329741	4736292

**APRIL 1972** 

#### **APRIL 1972**

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYS TEM

TEST/QC

AIR VEHICLE

#### ON-SITE LABOR

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL AR S	LABOR + BURDEN \$
Q-1	58	1.5	314	3.029	951		951
Q-2	58	<b>.</b>					
Q-3	58 58	16.5	2654	3.975	10550		10550
Q-1	59	51.0	8658	3.138	27168		27169
Q-2	59						2/100
Q-3	59 50	151.5	26779	3.369	90227		90227
Q-4	29	200.0					
Q=1 Q=2	60 60	308.0	53335	3.345	178412		178412
Q-3	60	683.5	114848	3.231	371049		371049
Q-4	60						3/10//
Q-1	61	1576.5	269054	3.230	863983		868983
Q-2	61	1202 0					
Q-4	61	1393.0	252615	3.144	794318		794318
Q-1	62	1716.0	292941	3.231	946548		946548
Q-2	62						510540
Q-3	62	1870.5	314195	3.402	1068942		1068942
Q-4	02	• • • • •					
Q = 1 Q = 2	63 63	1684.5	287465	3.619	1040306		1040306
0-3	63	2054-0	345040	4 34 9	1445710		1//5310
Q-4	63	200 180	349040	7040	1402/15	,	1465719
Q-1	64	2323.0	396468	3.592	1424274		1424274
Q-2	64						1767217
Q-3	64	1138.0	200271	3.592	719378		719378
Q-4	64						
Q-1	65	564.0	97737	3.847	375991		375991
Q-2	65						
Q-3	65	136.5	22991	4.263	98004		98004
W-4	05						

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM		EM	TEST/QC 1 AI		AIR VEHICLE			
			ON-SITE L	ABOR				
		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	
Q = 1 Q = 2	66 66		84	3.833	322	-40	282	
Q-3	66		12	4.083	49	46	95	
TOT	AL	15668.0	2685461		9481191	6	9481197	

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 AIR VEHICLE

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLL AR S	LABOR + BURDEN \$	ENGR Matl
Q-1	58	838.5	140876	4.498	633697	621920	1255617	2347
Q-2 Q-3	58 58	4503.5	756519	4.301	3254152	2858265	6112417	127533
Q-1 Q-2	50 59 50	8097.0	1381891	3.908	5400427	4699639	10100066	378864
Q-3 Q-4	59 59	11590.5	2039976	3.912	7980488	7237688	15218176	307531
Q = 1 Q = 1	60 60	13646.5	2365214	4.102	9702203	8425198	18127401	1460731
Q = 2 Q = 3	60 60	18253.0	3066646	3.943	12091192	10519010	22610202	1068555
Q = 1 Q = 2	61 61	31450.0	5367312	3.777	20272640	17738504	38011144	2757954
Q-3	61 61	27676.5	5018668	3.707	18603600	19983809	38587409	3170330
Q - 1 Q - 2	62 62	30990.0	5289113	3.644	19275406	20424239	39699645	1122159
Q-3	62 62	27310.5	4588293	3.788	17378358	1933566 <b>3</b>	36714021	1509779
Q-1	63 63	22308.5	3807078	4.084	15547700	17171454	32719154	1434773
Q-2 Q-3	63 63	25199.5	4233655	3.722	15758865	19110234	34869099	2667529
Q = 1 Q = 1	64 64	24355.0	4156438	4.071	16921040	22251963	39173003	507280
Q = 2 Q = 3	64 64	13237.5	2329853	4.201	9786975	12369579	22156554	495889
Q = 4 Q = 1 Q = 2	65 65	6511.0	1128335	4.098	4624293	569 <b>7795</b>	10322088	616224
Q = 2 Q = 3	65 65	1749.5	293896	5.877	1727265	1612773	3340038	226751
Q-1	65 66	39.0	6965	5.168	35995	52276	88271	54888

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 AIR VEHICLE

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
Q-2 66 Q-3 66 Q-4 66 Q-1 67 Q-2 67 Q-3 67	-1.5	-276	3.283	-906	-829	-1735	29951
TOTAL	267754.5	45970452		178993390	190109180	3691 025 70	17939068

# **APRIL 1972**

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 AIR VEHICLE

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		MFG MATL	SUBC	TOTAL MATERIAL	MPC	W IND TUNNEL	OTHER COST	TOTAL D/C \$
Q = 1 Q = 2	58 58			2 34 7	127	170368	10939	181307
Q-3	58 58		119071	246604	8031	583528	207890	791418
$\hat{Q} = 1$	59 59		3531218	3910082	125666	610704	743098	1353802
Q-3 Q-4	59 59		22077595	22385126	629551	695957	1268555	1964512
0 - 1 0 - 2	60	400	22356322	23817453	1518661	338115	2169302	2507417
Q-3	60 60	15609	17805774	18889938	1199030	403089	1617664	2020753
0 - 1	61	1001899	22966514	26726367	975245	296523	2384200	2680723
Q-3 0-4	61 61	4707321	29255957	37133608	1503312	237221	1866953	2104174
$\hat{\mathbf{Q}} = 1$	62	4903221	25968919	31994299	1300120	207559	2322574	2530133
Q-3	62 62	5119756	21324489	27954024	1200133	177914	2300733	2478647
0 - 1	63 63	5800375	14438909	21674057	1325907	148254	1721443	1869697
Q = 2 Q = 3	63 63	<b>655</b> 906 <b>7</b>	6836961	16063557	1128736	118797	-142029	-23232
Q = 4 Q = 1 Q = 2	64 64	5706444	7491653	13705377	1691259	1	951347	951348
Q-2 Q-3	64 64	2336960	250099	3082948	1112254	-1	871332	871331
Q = 4 Q = 1	65 65	508486	68654	1193364	358819	1	625325	625326
Q = 2	65 65	440881		667632	119113		250132	250132
Q-1	65 66	5227		60115	12468	11740	17868	29608

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# **APRIL 1972**

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 AIR VEHICLE

	MFG MATL	SUBC	TOTAL MATER IAL	MPC	W IND TUNNEL	OTHER COST	TOTAL 0/C \$
Q-2 66 Q-3 66 Q-4 65 Q-1 67 Q-2 67 Q-3 67	-30023		-72	-7	4328		4328
TOTAL	37075623	194492135	249506826	14208425	4004098	19187326	23191424

## II-112

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> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 AIR VEHICLE

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		SUB			TOTAL
		TOTAL	G & A	IDWA	COST
Q-1 Q-2	58	1439358			1439398
Q-3	58	7158470			7158470
Q-4 Q-1	59 59	15489616		23214	15512830
Q-3	59 59	40197365		8848	40206213
Q-4 Q-1	- 59 - 60	<b>4597</b> 0932	815552	13335	46799819
Q-2 Q-3	60 60	44719923	912644	96298	45728865
Q-4 Q-1	60 61	68393479	1366557	4372691	74132727
Q-2 Q-3	61 61	79328503	1566569	5913127	86808199
Q-4 Q-1	61 62	75524197	1305193	2924005	79753395
Q-2 Q-3	62 62	£8346825	1159073	2610066	72115964
Q-4 Q-1	62 63	57588815	1011192	2496459	61096466
Q-2 Q-3	63 63	52038160	916757	3068199	56023116
ୟ−4 ସ−1	63 64	55520987	1268114	4803875	61592976
Q-2 Q-3	64 64	27223087	612412	1561617	29397116
Q-4 Q-1	64 65	12499597	341493	4727	12845817
Q-2 Q-3	65 65	4376915	119740	6740	12040011
Q-4 Q-1	65 66	100462	E70E	0149	4503404
	50	190402	2132		196197

APRIL 1972

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 1 AIR VEHICLE

		SUB Total	GEA	IDWA	TOTAL COST
Q-2 Q-3 Q-4 Q-1 Q-2 Q-3	56 66 67 67 67	2514	75		2589

TOTAL 656009245 11401106 27903210 695313561





#### TABULATION OF DATES AIR VEHICLE NO. 1

ACTIVITY	START	COMPLETE
TNTTTAL SUBASSEMBLY	5-1-60 8-5-60	-
ASSEM FWD UPP INTER FUS	8-25-61	4-27-62
ASSEM LANDING GEAR	9-1-61	12-20-61
ASSEM FWD FUSELAGE	9-22-61	6-1-62
ASSEM AFT FUSELAGE	10-6-61	3-16-62
ASSEM LWR FWD INTER FUS	10-6-61	6-15-62
ASSEM AFT INTER FUS	11-3-61	6-22-62
ASSEM WINGS	$\frac{11-7-61}{5}$	7-26-62
MINC IOINE GIVING	5 - 10 - 62	7-21-62
ENGINE INSTALLATIONS	1 - 10 - 03	12 - (-03)
FIGTUE TINDIADIATIOND	4-0-02	TO-IS-OS (INITIAL CONFIG)
FINAL OPERATIONS	9-14-62	6-17-64
WING JOIN INNER WELD	1-7-64	2-18-64
WING JOIN OUTER WELD	1-20-64	4-25-64
ROLLOUT	-	5-11 <b>-</b> 64
GROUND VIBRATION TESTS	6-17-64	6-28-64
PROP & INTEG SYS TESTS	7-8-64	8-3-64
TAL TESTS	8-9-64	9-14-64
JOU NVOR J DILGUM	-	9-21-64
IST MACH 2 FLICHT	-	10-12-64
IST MACH 3 FLIGHT	-	3 - 24 - 07 10 - 1h - 65
		TO-T+-0)



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XB-70A AIR VEHICLE NO. 2 MILESTONES

WBS 1.0





## TABULATION OF DATES AIR VEHICLE NO. 2

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ACTIVITY	START	COMPLETE
	8 of (1	
TNITTAL FADALCAILON		-
THITTAT SOBADSENDIT		-
ACCENT LANDING GEAD	(-1-63	6-8-64
ASSEM LANDING GEAR	-	9-1-64
ASSEM FWD FUSELAGE	6-10-63	12-6-63
ASSEM AFT FUSELAGE	6-10-63	5 <b>-</b> 5 <b>-</b> 64
ASSEM LWR FWD INTER FUS	4-15-63	4 <b>-</b> 25 <b>-</b> 64
ASSEM AFT INTER FUS	10-18-63	5 <b>-2-</b> 64
ASSEM WINGS	6-14-63	7-3-64
FUSELAGE JOINING	4-12-64	9-14-64
WING JOINT SIZING	11-12-64	12-7-64
WING TO FUS JOINING	12-8-64	2-21-65
FINAL OPERATIONS	3-31-65	7-6-65
ENGINE INSTALLATIONS	3-25-65	4-22-65
ROLLOUT		5-29-65
GROUND VIBRATION TESTS	(NOT APPI	LICABLE)
PROP & INTEG SYS TESTS	6-23-65	6-29-65
TAXI TESTS	7-10-65	7-10-65
LST FLIGHT		7-17-65
LST MACH 1 FLIGHT	_	7-17-65
IST MACH 2 FLIGHT	_	9-29-65
LST MACH 3 FLIGHT	_	1-3-66







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E DOUT FRAME









EL TANK #8 -<u>WL-53.5+</u> -INSTENSCOTT STATIST AV. 2

-WL-60.94

- MITROSEN BOTTLE



TUSTED YENTATION FAIRNA AILE

VE ACCESS DOOR

TANK	#1 JP-6	7526
	#2 JP-6	1889
	#3 JP-6	54/3
	44 JP-6	6/19
	#5 JP-6	DRY
	16 JP-6	11,857
	#7 JP-6	.3290
	48 JP.6	4077

MAX USABLE FUEL CAPACITY AIY "

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MAX. USABLE FUEL CAPACITY

TOTAL

USABLE FUEL

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VERTICAL STREAMLER ACTUATOR

POLDOUT FRAME

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	FUOR	QUALT	rs FY

-	• • • • • • • • •
TANK T JP-G	7422
*2 JP-6	4900
3. JP-6	-5485
4 SP-6	5906
5 JP-6	: 4641
4. JP-6	10.787
1 V P-6	3/46
3 JP-6	3684
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A/V"1

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FUEL AIN'Z 45,971 GALLONS TOTAL USABLE






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AREA	6297
ASPECT EATIO	
TAPER PATIO	
SWEEP EACE 1257. :	elever of
AIRFOIL STRAIS-	14-58 LASUG
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30- 7 461 111	SIFIFO

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TOTAL ELEVENT AZEA-MET OF -	15 UNE 395
ELENCY AFEA FEE SOLL	137
ELEVON DEFLECTION	30°073'

HORIZONTAL STABILIZER	2
AREA (TOTAL)	45 54
ASPECT RATIG (TSTAL)	
TAPER RATIO (TSTAL)	
ROT CHORD (TOTI)	
TIP CHORD	
SNEEP BACK 252 ELEVEN	15)
AIRFER	30-70 HES
RAP AREA (TOTAL)	K.
FLAP DERECTION	
4 GZS @ EPSIS TO. GZS.	COTP.30 TONELN
MAC. LENGTH (TSTRE)	
DEFLECTION	4

RTICAL STABILIZER	
TOTAL VERTICAL FREA	4679:
SINGLE VERTICES FREA	233 24
ASPECT RATIO	
TAPER CATIO	·····
SWEEP BACK '257 ELEVENT)	
ROOT CHERD	
TIP CHORD	
AIR FOIL .	27.771511
40 0315 & REST & . CZ5 DTIP .	
MAC. LEIIGTN	1:
RUDDER AREA ( TTAL)	222222
RUDDER DEFLECTION ISTAR 2014	44101
RUDGER GEFLESTICN (GEAR UP)	imar

FOLOMIG TIP.	
AREA (EACH)	521.54
ASPECT RATIS	
TAPER RATIO	
DEFLECTION	25-2







FOLDOUT FRAME

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#### WORK BREAKDOWN STRUCTURE

#### PROGRAM TECHNICAL SUPPORT

WBS CODE: 2.0

	WBS	LE	VELS	
4	5	6	7	8

# 2.0 PROGRAM TECHNICAL SUPPORT

#### 2.20 DESIGN SUPPORT

- 2.20.1 Structural Dynamics
- 2.20.2 Loads
- 2.20.3 Stress
- 2.20.4 Weight Control
- 2.20.5 Structural Sciences
- 2.20.6 Lines and Dimensions
- 2.20.7 Materials and Processes
- 2.20.8 Non Metallics
- 2.20.9 External Aero
- 2.20.10 Internal Aero
- 2.20.11 Flight Sciences
- 2.20.12 Thermal Dynamics
- 2.20.13 Systems and Tactics
- 2.20.14 Design Support
- 2.20.15 Specifications
- 2.20.16 Design Producibility
- 2.20.17 Life Sciences

#### 2.21 OTHER R&D TESTING

- 2.21.1 Tests
  - 2.21.1.1 Structural R&D

SD72-SH-0003



# WORK BREAKDOWN STRUCTURE

#### PROGRAM TECHNICAL SUPPORT

WBS CODE: 2.0

	WBS	LEVEI	S		
4	5	6	7	8	
			2.2	1.1.2	Thermal Labs
			2.2	1.1.3	Material and Process Lab
			2.2	1.1.4	Lab Services
			2.2	1.1.5	Structural Dynamic Lab
			2.2	2.1.6	Wind Tunnels
			2.2	1.1.7	Hydraulic Lab
			2.2	1.1.8	Electrical Power Lab
			2.2	1.1.9	Flight Simulator
		2.21	•2	Model	Shop
	2.22	PROC	FRAM	MANAG	EMENT
		2.22	.1	Projec	ts/Administration
		2.22	•2	Advanc	e System Aeroprojects
		2.22	•3	Planni	ng
		2.22	•4	Compon	ent Controls
		2.22	•5	Data C	ontrol
		2.22	•6	Presen	tations
	2.23	LOG	ISTI	CS SUP	PORT
		2.23	.1	Field	Service
		2.23	•2	Mainte	nance Analysis
		2.23	•3	Mainte	nance Schedules
		2.23	.4	Packag	ing Studies



#### WORK BREAKDOWN STRUCTURE

# PROGRAM TECHNICAL SUPPORT

WBS CODE: 2.0

	WBS	LE	VELS	
4	5	6	7	8

## 2.24 WEAPON SYSTEM

2.24.1	Weapon System Analysis
2.24.2	Combat Effectiveness
2.24.3	Operational Analysis
2.24.4	Preliminary Analysis
2.24.5	Program Analysis
2.24.6	Preliminary Design
2.24.7	Advance Design



#### PROGRAM TECHNICAL SUPPORT WBS 2.0

#### DESCRIPTION

This WBS item has been established to collect those elements of the B-70 program which are closely related to the technical effort but are not identified to a particular vehicle subsystem. This item also contains all in-house and subcontractor costs associated with the design, development, fabrication and testing of the weapon system. The WBS level 5 elements comprising Program Technical Support 2.0 are:

2.20 2.21 2.22 2.23 2.24	Design Support Other R&D Testing Program Management Logistics Support Weapon System	<del>\$ \$ \$ \$ \$</del>	0 0 12,110,286 2,717,771 136,524,947	Page         II-166           Page         II-168           Page         II-174           Page         II-184           Page         II-193
Total	Cost	\$	151,353,004	Page II-131

It should be noted that Design Support 2.20 and other R&D Testing 2.21 contain no cost dollars. These items are used to provide a recap, for information purposes only, of the hours expended by each of the engineering groups defined by the WBS items 2.20 and 2.21. The North American Accounting system provides the ability to assign these groups to a particular subsystem, therefore, the actual costs are included in the subsystems (1.1 through 1.11.)

Explanation and definition of the costs associated with all the WBS level 5 items in Program Technical Support can be located on the pages indicated above. NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

# 4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT

			5-SUBSYS	5-SUBSYS	5-SUBSYS	
			22	23	24	TOTAL
			HOURS	HOURS	HOURS	HOURS
			DOLLARS	DOLLARS	DOLLARS	DOLLARS
DESIGN/ENGINEE	RING		1270491	257696	760698	2288885
LABOR AT \$	4.762		6216096	1173517	3510080	10899693
ENGR BURDEN	AT \$	4.248	5223627	1302642	3195969	9722238
PRODUCTION					220672	220672
LABOR AT \$	3.646				804564	804564
SHOP SUPPORT			16626	18407	14410	49443
LABOR AT \$	4.742		106311	84244	43901	234456
PLANN ING					101	101
LABOR AT \$	4.495				454	454
TEST/QC					1503	1503
LABOR AT \$	3.627				5452	5452
MFG BURDEN	AT \$	1.960	23954	99385	409156	532495
ENGR MATERIAL			3911	6808	50351	61070
MEG MATERIAL					5575	5575
SUBCONTRACT					122190126	122190126
MPC			475	578	4608407	4609461
OTHER COST			367573	3274	287993	658940
SUB-TOTAL			11942048	2670448	135112028	149724524
GEN & ADMIN			168238	47323	1412919	1628480
TOTAL COST			12110286	2717771	136524947	151353004

SUBDIVISION	OF WORK	:			
COST DETAIL	- SEE PAGE	II-176	II <b>-</b> 185	II-198	TT-132

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT

			DESIGN ZENGR HOURS DOLLARS	PROD HOURS DOLLARS	TOOLING AND STE HOURS DOLLARS	TEST /QC HOURS DOLLARS
DESIGN/ENGINEER	RING		2265365			37570
LABOR AT \$	4.762		10800714			23320
ENGR BURDEN	AT \$	4.248	9622103			100135
PRODUCTION				220672		
LABOR AT \$	3.645			804564		
SHOP SUPPORT			36439			13004
LABOR AT \$	4.742		195048			39408
PLANN ING				48		53
LABOR AT \$	4.495			163		291
TEST/QC			131	1084		288
LABOR AT \$	3.627		423	4216		813
MFG BURDEN	AT \$	1.960	129119	42262		361114
ENGR MATERIAL			11894			49176
MFG MATERIAL				5575		
SUBCONTRACT			36323328	85854040	12758	
MPC			1060028	3544765	589	4079
OTHER COST			658827			113
SUB-TOTAL			58801434	90255585	13347	654108
GEN & ADMIN			485162	1131495	174	11649
TOTAL COST			59286646	<b>91387</b> 080	13521	665757

TIME-PHASED COST				
DETAIL - SEE PAGE	II <b>-</b> 134	II-142	II <b>-</b> 147	II <b>-</b> 148

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT

			TOTAL
			HOURS
			DOLLARS
DESIGN/ENGIN	EERING		2288885
LABOR AT	\$ 4.762		10899693
ENGR BURDEN	AT \$	4.248	9722238
PRODUCTION			220672
LABOR AT	\$ 3.646		804564
SHOP SUPPORT			49443
LABOR AT	\$ 4.742		234456
PLANN ING			101
LABOR AT :	\$ 4.495		454
TEST/QC			1503
LABOR AT	\$ 3.627		5452
MFG BURDEN	AT \$	1.960	532495
ENGR MATERIAL	L.		61070
MFG MATERIAL			<b>5575</b>
SUBCONTRACT			122190126
MPC			4609461
OTHER COST			658940
SUB-TOTAL			149724524
GEN & ADMIN			1628480
TOTAL COST			151353004

TIME-PH	IA:	SED	COST		
DETAIL	-	SEE	PAGE	•	II <b>-</b> 154

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING 4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT SUBD OF WORK DESIGN/ENGINEERING

		MAN- Months	LABOR HOUR S	LABOR Rate	LABOR DOLLARS	BURDEN DOLL ARS	LABOR + BURDEN \$
Q-1	58	457.5	76930	4.509	346867	350 029	696896
Q-2	58						
Q-3	58	1049.0	176270	4.280	754512	691272	1445784
Q-4	58						
Q-1	59	1351.5	230678	4.239	977832	812548	1790380
Q-2	59						
Q-3	59	1486.0	261490	4.143	1083470	943378	2026848
Q-4	59						
Q-1	60	1163.5	201636	4.325	872097	761642	1633739
Q-2	60						
Q-3	<b>6</b> 0	802.5	134884	4.600	620470	488341	1108811
Q-4	60						1100011
Q-1	61	1721.5	293773	4.554	1337956	873647	2211603
Q-2	61					010011	444000
Q-3	61	939.0	170302	4.899	834320	753174	1587494
Q-4	61						1001104
Q-1	62	672.0	114748	5.196	596250	519012	1115262
Q-2	62					227026	AL ADEUE
Q-3	62	664.0	111513	5.023	560117	562404	1122521
Q-4	62					202.101	******
Q-1	63	660.0	112657	5.475	616802	611202	1228004
Q-2	63				*****	011202	1220004
Q-3	63	812.0	136411	5.721	780448	741026	1521474
Q-4	63		6		100170	2011F1	4751414 9
Q-1	64	724.5	123768	5-611	694432	720093	1414525
Q-2	64			20011	071132	120075	1414767
0-3	64	360.0	63348	5.918	374917	404470	770297
0-4	64				31 Y 24 Y	01110	117301
0-1	65	175.5	30434	6-266	190703	206557	207260
0 - 2	65		,	0 * L V V	L / U I U D	200331	371200
Q-3	65	90.0	15139	6.113	92545	99030	191575

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING 4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT SUBD OF WORK DESIGN/ENGINEERING

	MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-4 65						
Q-1 66	40.5	6966	5.778	40253	51953	92206
Q-2 66						12200
Q-3 66	26.5	4412	6.057	26723	32317	59040
TOTAL	13195.5	2265365		10800714	9622103	20422817

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### SHOP SUPPORT 4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT SUBD OF WORK DESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1	58	1.5	290	3.631	1053	981	2034
Q-2	58						
Q-3	58		-64	2.078	-133	-140	-273
Q-4	58						
Q-1	59		77	2.688	207	251	458
Q-2	59						
Q-3	59	6.0	1013	2.962	3000	4141	7141
Q-4	59	• •					
Q-1	60	-4.5	-690	2.984	-2059	-1866	-3925
Q-2	60						
Q-3	60					-59	-59
Q-4	60						
Q-1	61	4.5	804	3.041	2445	2384	4829
Q-2	61						
Q-3	61					240	240
Q-4	61						
0-1	62	19.5	3233	5.893	19053	6184	25237
Q-2	62						
Q-3	62	28.5	4839	7.108	34396	4765	39161
Q-4	62						
Q-1	6.5	114.0	19341	4.972	96154	86910	183064
Q-2	63						
Q-3	63	45.0	7596	5.389	40932	25328	66260
тот	AL	214.5	36439		195048	129119	324167

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM		TEST/QC	PROGRAM	TECHNICAL	SUPPORT
SUBD OF WO	)RK I	DESIGN/ENGI	NEERING	;	

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	59						
Q-2	59						
Q-3	59		35	2.829	9.9		00
Q-4	59				18 - 18 1		
Q-1	60		3	5.667	17		17
Q-2	60				<b>.</b> .		<b>T</b> 1
Q-3	60						
Q-4	60						
Q-1	61		92	3.304	304		304
Q-2	61						504
Q-3	61		1	3.000	3		3
Q-4	61						و
Q-1	62			4			
Q-2	62						
Q-3	62		5	7.200	36		36
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	63		- 5	7.200	-36		-36
тот	AL		131		423		423

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYS	TEM		2	PROGRAM	TECHNICAL	SUPPORT
SUBD	CF	WORK	DESIG	V/ENGINEERING	;	

		MAN-	LABOR	LABOR	LABOR	BURDEN	LABOR +	ENGR
		MONTHS	HOURS	RATE	DOLLARS	DOLLARS	BURDEN \$	MATL
Q-1 Q-2	58 58	459.0	77220	4.506	347920	351010	698930	458
Q-3 Q-4	58 58	1049.0	176206	4.281	754379	691132	1445511	-33
Q-1 Q-2	59 59	1351.5	230755	4.238	978039	812799	1790838	-77
Q-3 0-4	59 59	1492.0	262538	4.139	1086569	947519	2034088	1952
0-1 0-2	60 60	1159.0	200949	4.330	870055	759776	1629831	-1562
0-3 0-4	60 60	802.5	134884	4.600	620470	438282	1108752	-167
Q-1 Q-2	61 61	1726.0	294669	4.550	1340705	876031	2216736	7251
Q-3 Q-4	61 61	939.0	170303	4.899	834323	753414	1587737	3106
Q-1 Q-2	62 62	691.5	117981	5.215	615303	525196	1140499	129
Q-3 Q-4	62 62	692.5	116357	5.110	594549	567169	1161718	107
Q-1 Q-2	63 63	774.0	131998	5.401	712956	698112	1411068	
Q-3 Q-4	63 63	857.0	144002	5.704	821344	766354	1587698	
Q-1 Q-2	64 64	724.5	123768	5.611	694432	720093	8 1414525	
Q-3 Q-4	64 64	360.0	63348	5.918	374917	404 470	779387	642
Q-1 Q-2	65 65	175.5	30434	6.266	190703	206557	397260	88
Q-3 0-4	65 65	90.0	15139	6.113	92545	<del>9</del> 9030	191575	1
Q-1	<b>6</b> 6	40.5	6966	5.778	40253	51953	92206	

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT SUBD OF WORK DESIGN/ENGINEERING

	MAN- MON THS	LABOR HOUR S	LABOR	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$	ENGR MATL
Q-2 66			1				
Q-3 66	26.5	4412	6.057	26723	32317	59040	
TOTAL	13410.0	2301935		10996185	9751222	20747407	11894

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# 4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT

SUBD OF WORK DESIGN/ENGINEERING

			TOTAL		OTHER	SUB		TOTAL
		SUBC	MATERIAL	MPC	COST	TOTAL	G & A	COST
Q - 1	58 52		458	25	1322	700735		700735
Q-2 Q-3	53 53	7667900	7667867	67998	17498	9198874		9198874
Q-4 Q-1	58 59 50	15373379	15373302	407634	124559	17696333		17696333
Q-2 Q-3	59 59 50	2813500	2815452	77043	65035	4991618		4991618
Q=4 Q=1 Q=2	59 60 60	1000807	999245	59172	76573	2764821	52679	2817500
Q-3 Q-4	60 60	1289784	1289617	76504	53870	2528743	48180	<b>257</b> 6923
Q - 1 Q - 2	61 61	1005368	1012619	29499	87526	3346380	62187	<b>34</b> 0856 <b>7</b>
Q = 3 Q = 4	61 61	1140320	1143426	32936	46603	2810702	52232	2862934
Q-1 Q-2	62 62	1347123	1347252	42 82 4	19133	2549708	42797	2592505
Q-3 Q-4	62 62	1347122	1347229	42784	69282	2621013	43993	2665006
Q-1 Q-2	63 63	2080310	2080310	88388	56143	3635909	60792	3696701
Q-3 Q-4	63 63	359054	359054	11539	13486	1971777	32968	2004745
Q-1 Q-2	64 64	898661	898661	123422	21542	2458150	52304	° 2510454
Q-3 Q-4	64 64		642	234	1826	782089	16642	798731
Q-1 Q-2	65 65		88	26	2468	399842	10667	410509
Q-3 Q-4	65 65				1498	193073	5152	198225
Q-1	66				71	92277	2779	<b>9</b> 5056

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# 4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT SUBD CF WORK DESIGN/ENGINEERING

	SUBC	TOTAL MATERIAL	MPC	OTHER Cost	SUB TOTAL	G&A	TOTAL Cost
Q-2 66			· •				
Q-3 66				392	59432	1790	61222
TOTAL	36323328	36335222	1060028	658827	58801484	485162	59286646

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

TOTAL 1264.5 220672

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PRODUCTION 4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT SUBD OF WORK PRODUCTION

#### ON-SITE LABOR

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q-3 Q-4	59 59	576.0	101470	3.689	374296		374296
Q-1 Q-2 Q-3 Q-4	60 60 60 60	625.5	108474	3.643	395211		395211
0-1 0-2	61 61	63.0	10648	3.239	34484	39347	73831
Q-3 Q-4 Q-1 Q-2 Q-3 Q-4 Q-1 Q-2 Q-3	61 62 62 62 62 63 63 63		80	7.163	573	2847	3420

804564 42194 846758

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### PLANNING 4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT SUBD OF WORK PRODUCTION

	MAN- MONTHS	LABOR HOURS	LABOR	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1 61		48	3.396	163	68	231
TOTAL		48		163	68	231

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	TEST/QC			
4-SYSTEM	2	PROGRAM	TECHNICAL	SUPPORT
SUBD OF WORK	PRODUCTION			

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-1 Q-2	60 60		112	7.241	811		811
Q-3 Q-4	60 60						
Q-1	61	6.0	972	3.503	3405		3405
тот	AL	6.0	1084		4216		4216

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	4-SYSTEM 2 SUBD OF WORK PRODUCTION		PROGRAM TECHNICAL SUPPORT					
		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$	MFG Matl
Q-3 Q-4	59 59	576.0	101470	3.689	374296		374296	
Q-1 Q-2 Q-3 Q-4	60 60 60	625.5	108586	3.647	396022		396022	
Q-1 Q-2	61 61	69.0	11668	3.261	38052	39415	77467	4780
Q-3 Q-4 Q-1 Q-2 Q-3 Q-4 Q-1 Q-2 Q-3	61 61 62 62 62 62 63 63 63		80	7.163	573	2847	3420	795
тот	AL	1270.5	221804		808943	42262	851205	5575

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 SUBD OF WORK PRODUCTION

PROGRAM TECHNICAL SUPPORT

		SUBC	TOTAL MATERIAL	MPC	SUB Total	G & A	TOTAL COST
Q-3 Q-4	58 58	403574	403574	3581	407155		407155
Q-1 Q-2	59 59	3050417	3050417	80884	3131301		3131301
Q-3 Q-4	59 59	24671259	24671259	673913	25719468		25719468
Q-1 Q-2	60 60	18527001	18527001	1099276	20022299	381485	20403784
Q-3 Q-4	60 60	8625848	8625848	511770	9137618	174099	9311717
Q-1 Q-2	61 61	8648685	8653465	248197	8979129	166859	9145988
Q-3 Q-4	61 61	8536107	8536902	244778	8785100	163254	8948354
Q-1 Q-2	62 62	2991441	2991441	95073	3086514	51807	3138321
Q-3 Q-4	62 62	2991433	2991433	94986	3086419	51805	3138224
Q-1 Q-2	63 63	<b>479</b> 5488	4795488	203635	4999123	83585	5082 <b>7</b> 08
Q-3 Q-4	63 63	666689	666689	21463	688152	11506	699658
Q-1	64	1946098	1946098	267209	2213307	47095	2260402
TOT	AL	85854040	85859615	3544765	90255585	1131495	91387080

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SY \$	STEN	1	2		PROGRAM	TECHNICAL	SUPPORT
SUBD	OF	WORK	TOOLING	AND	STE		

		SUBC	TOTAL MATERIAL	MPC	SUB Total	GEA	TOTAL COST
Q-3 Q-4	59 59	4035	4035	110	4145		4145
Q-1 Q-2	60 60	4278	4278	253	4531	86	4617
Q-3 Q-4	60 60	3286	3286	194	3480	66	3546
Q-1 ( Q-2 (	61 61	1142	1142	32	1174	22	1196
Q-3	61	17	17		17		17
тоти	AL	12758	12758	589	13347	174	13521

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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TIME	PHASED EXP	END.
<b>B-7</b> 0	AIRCRAFT S	TUDY

#### DESIGN/ENGINEERING 4-SYSTEM 2 SUBD OF WORK TEST/QC PROGRAM TECHNICAL SUPPORT

$\begin{array}{c ccccccccccccccccccccccccccccccccccc$			MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	Q-1	60						
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	Q-2	60						
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	Q-3	60		1	3.000	3		3
Q-16176.5130344.400573524430510165 $Q-2$ 615.59513.8433655104171407 $Q-4$ 610-16228.549493.97819687212244091 $Q-2$ 620-2620-2620-2620-262 $Q-3$ 6216.528883.60410409136382404 $Q-4$ 620-16311.519194.4988632104221905 $Q-2$ 630-463-1.5-2233.399-758128-63 $Q-4$ 631.999-11110511 $Q-1$ 641.999-1111	Q-4	60				2		<u> </u>
Q-2615.59513.8433655104171407 $Q-4$ 61 $Q-4$ 61 $Q-4$ 61 $Q-4$ 62 $Q-2$ 62 $Q-2$ 62 $Q-2$ 62 $Q-2$ 62 $Q-3$ 6216.528883.60410409136382404 $Q-4$ 62 $Q-4$ 62 $Q-4$ 62 $Q-2$ 63 $Q-4$ 63 $Q-2$ 63 $Q-4$ 63 $Q-2$ $Q-$	Q-1	61	76.5	13034	4.400	57352	44 305	101657
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	Q-2	61						201051
Q-4 $61$ $Q-1$ $62$ $28.5$ $4949$ $3.978$ $19687$ $21224$ $4091$ $Q-2$ $62$ $16.5$ $2888$ $3.604$ $10409$ $13638$ $2404$ $Q-4$ $62$ $11.5$ $1919$ $4.498$ $8632$ $10422$ $1905$ $Q-2$ $63$ $-1.5$ $-223$ $3.399$ $-758$ $128$ $-63$ $Q-4$ $63$ $1$ $.999$ $-1$ $1$ $Q-1$ $64$ $1$ $.999$ $-1$ $1$	Q-3	61	5.5	951	3.843	3655	10417	14072
Q-1 $62$ $Q-2$ $28.5$ $4949$ $3.978$ $19687$ $21224$ $4091$ $Q-2$ $62$ $Q-4$ $16.5$ $2888$ $3.604$ $10409$ $13638$ $2404$ $Q-4$ $62$ $Q-4$ $63$ $11.5$ $1919$ $4.498$ $8632$ $10422$ $1908$ $Q-2$ $63$ $Q-2$ $-1.5$ $-223$ $3.399$ $-758$ $128$ $-63$ $Q-4$ $63$ $Q-1$ $1$ $.999$ $-1$ $1$	Q-4	61						
Q-2 $62$ $Q-3$ $62$ $Q-4$ $62$ $Q-4$ $62$ $Q-1$ $63$ $Q-2$ $63$ $Q-3$ $63$ $Q-3$ $63$ $Q-3$ $63$ $Q-4$ $62$ $Q-2$ $63$ $Q-3$ $63$ $Q-4$ $63$ $Q-4$ $63$ $Q-1$ $64$ $1$ $999$ $-1$ $1$	Q-1	62	28.5	4949	3.978	19687	21224	40911
Q-3 $62$ $16.5$ $2888$ $3.604$ $10409$ $13638$ $2404$ $Q-4$ $62$ $Q-1$ $63$ $11.5$ $1919$ $4.498$ $8632$ $10422$ $1905$ $Q-2$ $63$ $-1.5$ $-223$ $3.399$ $-758$ $128$ $-63$ $Q-4$ $63$ $1$ $.999$ $-1$ $1$ $Q-1$ $64$ $1$ $.999$ $-1$ $1$	<b>Q-</b> 2	62						
Q-4 $62$ $Q-1$ $63$ $11.5$ $1919$ $4.498$ $8632$ $10422$ $1905$ $Q-2$ $63$ $-1.5$ $-223$ $3.399$ $-758$ $128$ $-63$ $Q-4$ $63$ $0-4$ $63$ $1$ $.999$ $-1$ $1$ $I$ $I$ $I$ $I$ $I$ $I$ $I$ $I$	Q-3	62	16.5	2888	3.604	10409	13638	24047
Q-1       63       11.5       1919       4.498       8632       10422       1905 $Q-2$ 63 $Q-3$ 63 $-1.5$ $-223$ $3.399$ $-758$ 128 $-63$ $Q-4$ 63 $Q-1$ 64       1 $.999$ $-1$ 1	Q-4	62						
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	Q-1	6.3	11.5	1919	4.498	8632	10422	19054
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	Q-2	63						
Q-4 63 Q-1 64 1 .999 -1 1	Q-3	63	-1.5	-223	3.399	-758	128	-630
Q-1 64 1 .999 -1 1	Q-4	63						
	Q-1	64		1	• 99 9	-1	1	
TUTAL 137.0 23520 98979 100135 19911	тот	AL	137.0	23520		98979	100135	199114

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### SHOP SUPPORT 4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT SUBD OF WORK TEST/QC PROGRAM TECHNICAL SUPPORT

#### ON-SITE LABOR

		MAN- MON THS	LABOR HOUR S	LABOR	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q-3	58 50	4.5	669	2.709	1812	2588	4400
Q-1 Q-2	58 59 59	4.5	886	2.837	2514	3359	5873
Q-3 Q-4	59 59	6.0	993	2.763	2744	4122	6866
Q-1 Q-2	60 60	10.5	1759	2.895	5093	12203	17296
Q-3 0-4	60 60	6.0	914	2.745	2509	187088	189597
Q-1	61 61	35.0	5998	3.167	18996	99759	118755
Q-3	61	2.5	411	3.416	1404	46572	47976
Q = 1 Q = 1	62 62		122	2.533	309	443	752
Q-3	62 62	4.5	677	3.065	2075	2685	4760
Q-1	63 (2	1.5	176	3.295	580	715	1295
Q-2 Q-3	63 (7	3.0	401	3.426	1374	1581	2955
Q-4 Q-1	64 64		-2	1.000	-2	-1	-3
тот	AL	78.0	13004		39408	361114	400522

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	PLANNING			
4-SYSTEM SUBD OF WORK	2 TEST/QC	PROGRAM	TECHNICAL	SUPPORT

		MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q = 1	63 63		28	3.357	94	111	205
Q-3	63		25	7.880	197	-111	86
тот	AL		53		291		291

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 2 SUBD OF WORK TEST/QC

PROGRAM TECHNICAL SUPPORT

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 58		44	2.659	117		117
Q = 4 53 Q = 1 59 Q = 2 59		24	3.333	80		80
Q-3 59 Q-4 59		29	2.966	86		86
Q-1 60 Q-2 60		4	2.750	11		11
Q-3 60 Q-4 60						
Q - 1 61 Q - 2 61	1.0	135	2.741	370		370
Q-3 61 Q-4 61		16	2.500	40		40
Q-1 62 Q-2 62		-3	2.000	-6		-6
Q-3 62 Q-4 62		28	2.786	78		78
Q-1 63 Q-2 63						
Q-3 63 Q-4 63		9	3.889	35		35
0-1 64		2	1.000	2		2
TOTAL	1.0	288		813		813

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 SUBD OF WORK TEST/QC

PROGRAM TECHNICAL SUPPORT

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$	ENGR MATL
Q-3 Q-4	53 58	4.5	713	2.705	1929	2588	4517	2199
Q-1 Q-2	59 59	4.5	910	2.851	2594	3359	5953	590
Q-3 Q-4	59 59	6.0	1022	2.769	2830	4122	6952	239
Q-1 Q-2	60 60	10.5	1763	2.895	5104	12203	17307	-900
Q - 3 Q - 4	<b>6</b> 0	6.0	915	2.745	2512	187088	189600	146
Q-1 Q-2	61 61	112.5	19167	4.003	76718	144064	220782	12864
Q-3 Q-4	61 61	8.0	1378	3.700	5099	56989	62088	8835
Q-1 Q-2	62 62	28.5	5068	3.944	19990	2 <b>16</b> 6 <b>7</b>	41657	20574
Q-3 Q-4	62 62	21.0	3593	3.496	12562	16323	28885	-1191
$\hat{Q} = 1$ $\hat{Q} = 2$	63 63	13.0	2123	4.383	9306	11248	20554	6109
Q-3 Q-4	63 63	1.5	212	4.000	848	1598	2446	-281
Q-1	64		1	• 999	-1		-1	- 8
тот	AL	216.0	36865		139491	461249	600740	49176

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 SUBD OF WORK TEST/QC

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PROGRAM TECHNICAL SUPPORT

	МРС	OTHER COST	SUB Total	G&A	TOTAL COST
Q-3 58 Q-4 58	120		6836		6836
Q-1 59 Q-2 59	50		6593		6593
Q-3 59 Q-4 59	20		7211		7211
Q-1 60 Q-2 60	-118		16289	310	16599
Q-3 60 Q-4 60	19		189765	3616	193381
Q-1 61 Q-2 61	1087		234733	4362	239095
Q-3 61 Q-4 61	746	113	71782	1334	73116
Q-1 62 Q-2 62	1621	3	63855	1072	64927
Q-3 62 Q-4 62	-93	19	27620	464	28084
Q-1 63 Q-2 63	601		27264	455	27719
Q-3 63 Q-4 63	27	-22	2170	36	2206
Q-1 64	-1		-10		-10
TOTAL	4079	113	654108	11649	665757

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING

## 4-SYSTEM

# PROGRAM TECHNICAL SUPPORT

		MAN-	LABOR	LABOR	LABOR	<b>BUR DEN</b>	LABOR +
		MUNIHS	HUURS	RATE	DOLLARS	DOLLARS	BURDEN \$
Q-1	58	457.5	76930	4.509	346867	350.029	696896
Q-2	58						0,00,0
Q-3	58	1049.0	176270	4.280	754512	691272	1445784
Q-4	58						2110101
Q-1	59	1351.5	230678	4.239	977832	812548	1790380
0-2	59						
Q-3	59	1486.0	261490	4.143	1083470	943.378	2026848
Q-4	59						2020010
Q-1	60	1163.5	201636	4.325	872097	761642	1633739
Q-2	60						
Q-3	60	802.5	134885	4.600	620473	488341	1108814
Q-4	60						1100011
Q-1	61	1798.0	306807	4.548	1395308	917952	2313260
Q-2	61						20200
Q-3	61	945.0	171253	4.893	837975	763591	1601566
Q-4	61						1001900
Q-1	62	701.5	119697	5.146	615937	540236	1156173
Q-2	62						
Q-3	62	681.0	114401	4.987	570526	576042	1146568
Q-4	62						11 10900
Q-1	63	671.5	114576	5.459	625434	621624	1247058
Q-2	63						
Q-3	63	811.0	136194	5.725	779690	741 162	1520852
Q-4	63						135003E
Q-1	64	724.5	123769	5.611	694431	720094	1414525
Q-2	64				· · · -		
Q-3	64	360.0	63348	5.918	374917	404470	779387
Q-4	64	•					
Q-1	65	175.5	30434	6.266	190703	206557	397260
Q-2	65						577200
0-3	65	90.0	15139	6.113	92545	99030	191575
0-4	65						

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

**APRIL 1972** 

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# 4-SYSTEM

DESIGN/ENGINEERING 2

# PROGRAM TECHNICAL SUPPORT

	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1 66 Q-2 66	40.5	6966	5.778	40253	51953	92206
Q-3 66	26.5	4412	6.057	26723	32317	59040
TOTAL	13335.0	2288885		10899693	9722238	20621931

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	4-SYSTEM		PRODUCTIC 2	)n Program	TECHNICAL	SUPPORT		
			ON-SITE	LABOR				
		MAN- MONTHS	LABO HOUR	DR LA LS R	BOR L Ate dol	LABOR LLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 Q-4	59 59	576.0	10147	'0 3 <b>.</b>	689 31	74296		374296
Q-1 Q-2 Q-3	60 60 60	625.5	10847	4 3.	643 39	95211		395211
Q-4 Q-1 Q-2	60 61 61	63.0	1064	8 3.	239 3	34484	39347	73831
Q-3 Q-4 Q-1 Q-2 Q-3 Q-4 Q-1 Q-2 Q-3	61 62 62 62 62 62 63 63 63		8	07.	163	573	284 <b>7</b>	3420
тот	AL	1264.5	22067	2	80	4564	42194	846758
# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SF
4-SYSTEM	2

SHOP SUPPORT

SISIEM

PROGRAM TECHNICAL SUPPORT

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1 58	1.5	290	3.631	1053	981	2034
Q-3 58 0-4 58	3.0	605	2.775	1679	2448	4127
Q-1 59 Q-2 59	6.0	963	2.826	2721	3610	6331
Q-3 59 Q-4 59	11.5	2006	2.863	5744	8263	14007
Q = 1 60 Q = 2 60	6.0	1069	2.838	3034	10337	13371
Q-3 60 Q-4 60	6.0	914	2.745	2509	187029	189538
Q-1 61 Q-2 61	40.0	6802	3.152	21441	102143	123584
Q-3 61 Q-4 61	2.5	411	3.416	1404	46812	48216
Q = 1 62 Q = 2 62	19.5	3355	5.771	19362	6627	25989
Q-3 62 Q-4 62	33.0	5516	6.612	36471	7450	43921
Q = 1  63 Q = 2  63	114.0	19517	4.956	96734	87625	184359
Q-3 63 Q-4 63	48.0	7997	5.290	42306	26909	69215
Q-1 64		-2	1.000	-2	-1	-3
TOTAL	291.0	49443		234456	490233	724689

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	PLANNING			
4-SYSTEM	2	PROGRAM	TECHNICAL	SUPPORT

#### ON-SITE LABOR

		MAN- MON TH S	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
0-1	61		48	3.396	163	68	231
Q-2	61						
Q-3	61						
Q-4	61						
Q-1	62						
Q-2	62						
Q-3	62						
Q-4	62						
Q-1	63		28	3.357	94	111	205
Q-2	63						205
Q-3	63		25	7.880	197	-111	86
тот	AL		101		454	68	522

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# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

		TEST/QC
4–SYST	EM	2

PROGRAM TECHNICAL SUPPORT

# ON-SITE LABOR

		MAN- MONTHS	LABOR	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	58		44	2.659	117		117
Q-4	58						** '
Q-1	59		24	3.333	80		80
Q-2	59						00
Q-3	59		64	2.891	185		185
Q-4	59				<b>,</b>		102
Q-1	60	1.0	119	7.050	839		839
Q-2	60						
Q-3	60						
Q-4	60						
Q-1	61	7.5	1199	3.402	4079		4079
Q-2	61						4019
Q-3	61		17	2.529	43		43
Q-4	61						12
Q-1	62		-3	2.000	-6		-6
Q-2	62				_		0
Q-3	62		33	3,455	114		114
Q-4	62						117
Q-1	63						
Q-2	63						
Q-3	63		4	• 249	-1		-1
Q-4	63				-		*
Q-1	64		2	1.000	. 2		2
тот	AL	8.5	1503		5452		5452

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# **APRIL 1972**

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# 4-SYSTEM 2

PROGRAM TECHNICAL SUPPORT

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$	ENGR MATL
Q-1 Q-2	58 58	459.0	77220	4.506	347920	351010	698930	458
Q-3 Q-4	58 58	1052.0	176919	4.275	756308	693720	1450028	2166
Q-1 Q-2	59 59	1357.5	231665	4.233	980633	816158	1796791	513
Q-3 Q-4	59 59	2073.5	365030	4.010	1463695	951641	2415336	2191
ଦ−1 ଦ−2	60 60	1796.0	311298	4.083	1271181	771979	2043160	-2462
Q-3 Q-4	60 60	808.5	135799	4.588	622982	675370	1298352	-21
Q-1 Q-2	61 61	1908.5	325504	4.471	1455475	1059510	2514985	20115
Q-3 Q-4	61 51	947.5	171761	4.890	839995	813250	1653245	11941
Q-1 Q-2	62 62	721.0	123049	5.163	635293	546863	1182156	20703
Q-3 Q-4	62 62	714.0	119950	5.061	607111	583492	1190603	-1084
Q-1 Q-2	63 63	785.5	134121	5.385	722262	709360	1431622	6109
Q-3 Q-4	63 63	859.0	144220	5.701	822192	767960	1590152	-281
Q-1 Q-2	64 64	724.5	123769	5.611	694431	720093	1414524	-8
Q-3 Q-4	64 64	360.0	63348	5.918	374917	404 470	779387	642
Q-1 Q-2	65 65	175.5	30434	6.266	190703	206557	397260	88
Q-3 Q-4	65 65	90.0	15139	6.113	92545	99030	191575	
Q-1	66	40.5	6966	5.778	40253	51953	92206	

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT

	MAN- MONTHS	L ABOR HOUR S	LABOR RATE	LABOR DCLLARS	BUR DEN DOLL AR S	LABOR + BURDEN \$	ENGR MATL
Q-2 66 Q-3 66	26.5	4412	6.057	26723	32317	59040	
TOTAL	14899.0	2560604		11944619	10254733	22199352	61070

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT

		MFG Matl	TCOL/STE MATL	SUBC	TOTAL MATERIAL	MPC	OT HE R COST	SUB TOTAL
Q-1	58				458	25	1322	700735
Q-2	58			0071/7/	0070440	71 (00	17/00	0(100/5
0-4	20 59			8071474	8013640	11099	17498	9012805
$\hat{\mathbf{Q}} = 1$	59			18423796	18424309	488568	124559	20834227
0-2	59			10123130				the lot of and the lot of
0-3	59			27488794	27490985	751086	65035	30722442
Q-4	59							
Q-1	60			19532086	19529624.	1158583	76573	22807940
Q-2	60							
Q-3	60			9918918	9918897	588487	53870	11859606
Q-4	60	( 7.0.0		0/55105	0/0000	270.015	0750/	106(1/1/
0 - 1	61	4780		5055195	9680090	278815	81526	12061410
Q = Z	61	795		9676444	9689180	278460	46716	11667601
Q-4	61			2010111	,00,100	210105	.0,10	11001001
Q-1	62			4338564	4359267	139518	19136	5700077
Q-2	62							
Q-3	62			4338555	4337471	137677	69301	5735052
Q-4	62			1075700	(001007	202/2/	F ( 1 / 2	0449904
	63			6815198	6881907	292024	20143	8002290
0-3	63			1025743	1025462	33029	13464	2662107
0-4	63			1020143	1020402	33627	19101	2002101
Q-1	64			2844759	2844751	390630	21542	4671447
Q-2	64							
Q-3	64				642	234	1826	782089
Q-4	64							
Q-1	65				88	26	2468	399842
0-2	65						1/00	102070
Q-3	65						1498	143013
6-4	65 44						71	97777
Q-1	00						11	76611

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# 4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT

	MFG Matl	TCOL/STE MATL	SUBC	TOTAL MATERIAL	MPC	OT HE R COS T	SUB Total
Q-2 66 Q-3 66						392	59432
TOTAL	5575		122190126	122256771	4609461	658940	149724524

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT

	GεA	TOTAL Cost
Q-1 58 Q-2 58		700735
Q-3 58		9612865
Q = 1 59 Q = 2 59		20834227
Q-3 59 Q-4 59		30722442
Q = 1 60 Q = 2 60	434560	23242500
Q-3 60 Q-4 60	225961	12085567
Q = 1  61 Q = 2  61	233430	12794846
Q-3 61 Q-4 61	216820	11884421
Q-1 62 Q-2 62	95676	5795753
Q-3 62 Q-4 62	96262	5831314
Q-1 63 Q-2 63	144832	8807128
Q-3 63 Q-4 63	44510	2706617
Q = 1 64 Q = 2 64	99399	4770846
Q-3 64 Q-4 64	16642	798731
Q-1 65 Q-2 65	10667	410509
Q-3 65 Q-4 65	5152	198225
Q-1 66	2779	95056

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 PROGRAM TECHNICAL SUPPORT

G	3	A	Т	OTAL Cost

Q-2	66		
Q-3	66	1790	61222

TOTAL 1628480 151353004



#### COST DEFINITION WBS 2.20

#### DESIGN SUPPORT

This WBS item is utilized to display, as information only, a matrix of the engineering groups and disciplines which normally support the development and design of the complete aircraft rather than an individual subsystem. No costs are contained in this item as the North American cost accounting system provides the ability to assign these groups to a vehicle subsystem. The assignment of the design support groups to the subsystems utilizing the cost accounting system detail was a joint discussion made by NR and NASA at the conclusion of Phase I of this study.

To provide the analysts with the ability to isolate and remove, if desired, one or all of these design support groups from the subsystems, the following matrix was developed. Each group associated with this WBS item has been identified and a recap of the hours charged to individual subsystems has been prepared. No cost data has been assigned to this item. The hours data is presented as information only.



# Space Division North American Rockwell

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#### WBS 2.20 DESIGN SUPPORT-ENGINEERING GROUP MATRIX

W	BS			WBS	WBS	WBS	WBS	WBS	WBS	WBS	WBS	WBS	SUBSYSTEM
α	ODE	GROUP	GROUP TITLE	1.0	1.1	1.2	1.3	1.4	1.5	1.6	1.7	1.9	TOTAL
2.	20.1	51	Structural Dynamics	421171									421171
2.2	20.2	73	Structural Loads	•	306050		-742	· •			•		306050
2.2	20.3	10	Structural Analysis		1019126		5662			8204			1032992
2.2	20.4	11	Weight Control	259843	32796		7559				1925		302123
2.2	20.5	33	Advanced Structures		256050	(							256050
		34	Structural Projects		242904	6515	-0-6			35779	2132		287330
2.2	20.6	30	Numerical Design		215538	06000	5816						221354
2.2	20.7	50	Metallurgy		228/2	86029	00717						195010
	·	54	Material & Process		409495		19053	39063		11016		2069	40(011
2.2	20.8	75	Non-Metallics	00060	131000			24500		11010		3000	1 10450
2.2	20.9	13	Aerodynamics	20003	02305		6659						82012
	İ	131	Aero Special Projects	5010	10301		610070						100521
		133	Aerodynamics	21240	90010 60385		04401						012/8
2.2	20.10	13	Aerodynamics	20003	02307		6)1)107						100532
		133	Aerodynamics	21240	90011		1222						13300
2.0		120	Flight Sciences	17872	32130	221086	100++		36131				307220
2.0		175	Fit Controls Analysis	11012	3608	221000			23278	237104			258460
0	20.12	61	Design Support	Thoreh	2035				2,5++0	7576	1570		160914
5	20.15	57	Fraineering Space	40810	23228	18350	32202	<b>ร</b> 8มรม		311110	11611	17726	236810
2.0	20.10	12	Checking	40019	146505	11297	12431	12895		16410	8084		207622
2."	20.10	52	Design Productibility		241403		8582	33252					283237
0	20 17	150	Idfa Sciences		242405		0,02	5,2,2			13406		13406
2.1	20.11	1,0	WIE DETENCED										25+00
						ahaann	20(828	2 (02 k)	5.0570	245600	00707	ograh	67970()
			WBS 2.20 TOTAL	1006/30	3477030	343277	320030	100144	עורעכ	347029	30131	20194	5101304
					]								· · .
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1						1							
				1									

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#### COST DEFINITION WBS 2.21

#### OTHER R&D TESTING

This WBS item is utilized to display, as information only, a matrix of the engineering lab groups, which normally support the development and design of the complete aircraft rather than an individual subsystem. No costs are contained in this item as the North American cost accounting system provides the ability to assign these groups to a vehicle subsystem. The assignment of the design support groups to the subsystems utilizing the cost accounting system detail was a joint decision made by NR and NASA at the conclusion of Phase I of this study.

To provide the analysts with the ability to isolate and remove, if desired, one or all of these design support groups from the subsystem, the following matrix was developed. Each group associated with this WBS item has been identified and a recap of the hours charged to individual subsystems has been prepared. No cost data has been assigned to this item. The hours data is presented as information only.



## WBS 2.21 OTHER R&D TESTING-ENGINEERING GROUP MATRIX

WBS CODE	GROUP	GROUP TITLE	WBS 1.0	WBS 1.1	WBS 1.2	WBS 1.3	WBS 1.4	WBS 1.5	WBS 1.6	WBS 1.7	WBS 1.9	SUBSYSTEM TOTAL
2.21.1.1 2.21.1.2 2.21.1.3 2.21.1.4 2.21.1.6 2.21.1.7 2.21.1.8 2.21.1.9	67 92 146 66 97 95 13 100 11 4	Structural Test Lab Thermodynamics Thermodynamics Metallic Materials Lab Laboratory Services Wind Tunnel Projects Wind Tunnel Projects Hydraulic Lab Electrical Power Lab Flight Simulation	5582 36764 10296 95149 7291	387804 336180 126386 138560 75897 40465 5532 71659	30465 18587 8977 7980	22691 19595 17776 34œ3 78321 32172	4854 17704 57963 167950 36@6 4803	24696 24499	39404 60007 178540	4118 5657 6334 14310 3810	5078 38493.	396776 53156 49421 371660 314929 163256 100396 357039 197490 298275
2.21.2	14	SUB-TOTAL Wind Tunnel Models	155082 246723	1182483 125534	66009	204578	289300	49195 261674	277951	34229	43571	2302398 633931
		TOTAL	401805	1308017	66009	204578	289300	310869	277951	34229	43571	2936329

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# WIND TUNNEL TESTS

WBS 2.21.1.6



# WIND TUNNEL MODEL SUMMARY

WBS 2.21.1.6

AERODYNAMIC TESTS	1	AIR INLET TESTS	1	VIBRATION & FLUTTER TE	STS
MODEL	SCALE	MODEL	SCALE	MODEL	SCALE
MODEL SMALL AERO FORCE SMALL AERO FORCE (-154 LINES) BLUFF STORE BLUFF STORE BLUFF STORE BLUFF STORE BLC TEST SPECIMEN COMPRESSION AFTERBODY PRELIMINARY TRISONIC FORCE WING & WEDGE FORCE MODEL I FORCE MODEL I FORCE MODEL III FORCE MODEL III FORCE MODEL IV FORCE MODEL IV FORCE MODEL V CANOPY DRAG ENCAPSULATED SEAT ENCAPSULATED SEAT 2 DIMENSIONAL WING SURFACE ROUGHNESS WING PRESSURE PRESSURE PROBE	.008 .008 .035 .06 .008 NONE .03 NONE .03 .03 .03 .008 .03 .008 .03 .008 .03 .008 .03 .008 .03 .008 .03 .008 .03 .008 .025 NONE NONE .024 1.00	PRELIM INLET DUCT RESEARCH PRELIM INLET DUCT RESEARCH INFINITELY VARIABLE DUCT BASE PRESSURE BASE PRESSURE POROUS MATERIAL RESEARCH SUPERSONIC DIFFUSER INLET CONTROL INLET CONTROL INLET CONTROL (HSD) INLET-ENGINE CONTROL INLET FORCE INLET DUCT - I INLET DUCT - I INLET DUCT - III INLET DUCT - IIIR	.04 .05 NONE .04 .045 NONE .04 .10 .25 .04 .577 .04 .05 .05 .05	FLOATING PANEL FLUTTER CENTER LINE VERTICAL COMPLETE MODEL COMPLETE MODEL COMPLETE MODEL VERTICAL STABILIZER FLUTTER WING FLUTTER WING VERTICAL TAIL CANARD SUSPENSION SYSTEM q REDUCER COMPLETE TRAN FUSELAGE FLUTTER PANELS FUSELAGE FLUTTER PANELS FORCE MODEL - IV INLET FLUTTER RAMPS	NONE .06 .10 .03 .04 .06 .04 .10 .15 .13 .20 .06 FULL .04 .89 FULL .04 .10

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Wind Tunnel Occupancy Hours





#### COST DEFINITION WBS 2.22

#### PROGRAM MANAGEMENT

Cost data assigned to this WBS item include the activities associated with the following functions:

- a) control of program costs and technical accomplishments
- b) phasing
- c) operational, maintenance and functional configuration analysis
- d) labor costs of personnel supporting engineering representation at associate contractors and subcontractors
- e) final preparation of reports and data for presentation to the customer
- f) program planning
- g) data control

The engineering groups providing support to these tasks are:

Group No.	Engineering Group Title	Hours Expended
2	Propulsion Design and Development	2,389
11	Weight Control	17,174
13	Aerodynamics	3,285
15	Data Control	17.204
19	Propulsion System Development	4,831
20	Operation Planning	44,726
27	Proposal and Procurement Planning	11,985
28	Field Service Publications	4,680
42	AGE Electrical Equipment	4,251
45	Design Illustrations	245,292
46	Mechanical and Propulsion Adm.	5,225
55	Flight Control Analysis	3,802
57	Engineering Specifications	75,671
59	X-Ray Photo Template Lab	17,561
61	Operations Administration	59,140
65	Design Support	2,792
60	Photo-Instrumentation	24,314
09	B- (O Planning	9,042
(1 70	Support Equipment AGE	8,175
(2	Engineering Computing	34,351
76	Flight Test Project Engineering	19,579
	Reliability and Maintainability	17,097
79	Packaging	10,113
00	Flight Operations	16,494
04	Weapon System Analysis	57,459
90	Electronic System Administration	7,811
94	Flight Simulation	21,060
98	Field Service Publications	3,745
103	Flight Test Engineering Adm.	16,778
107	Flight Test Maintenance Support	12,605
116	Advanced Aerospace Systems	7,429
117	Preliminary Analysis	4,169
119	Preliminary Design	13,429



Group No.	Engineering Group Title	Hours Expended
120	R&D Programming	8,574
121	Advanced Projects	15,839
123	Preliminary Design	4,150
125	Electrical System Equipment	7,233
127	B-70 Project Group	276,757
128	Operations Research	39,784
129	Operations and Military Systems	
	Analysis	58,555
169	B-70 Project Planning	18,736
Various		37,205

Total Engineering Hours

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1,270,491

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 2 5-SUBSYSTEM 22 PROGRAM MANAGEMENT

•

			DESIGN			
			DOLLARS	DOLLARS DOLLARS		
DESIGN/ENGINEERI	NG		1270491	1270491		
LABOR AT \$ 4	.893		6216096	6216096		
ENGR BURDEN	AT \$	4.112	5223627	5223627		
SHOP SUPPORT			16626	16626		
LABOR AT \$ 6	.394		106311	106311		
MFG BURDEN	AT \$	1.441	23954	23954		
ENGR MATERIAL			3911	, 3911		
MPC			476	476		
OTHER COST			367673	367673		
SUB-TOTAL			11942048	11942048		
GEN & ADMIN			168238	168238		
TOTAL COST			12110286	12110286		

TIME-PHASED COST DETAIL - SEE PAGE II-177

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

DE SIGN / ENGINEERING 4-SYSTEM 2 5-SUBSYSTEM 22 PROGRAM MANAGEMENT SUBD DF WORK DE SIGN / ENGINEERING

	MAN- MONTHS	L ABOR HOUR S	LABOR RATE	LABGR DGLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
0-1 54	<b>) 16 "7</b> 6"	2 ( 2 0 1				
		26391	4.406	116273	120078	236351
		70070				
Q-3 30	0 40 <b>2</b> €Ŭ	18019	4.320	337295	304342	641637
0-1 50	) ) 700 E	10/000	1 201			
Q = 1 25	Ø (90+0 N	1.5459.5	4.330	585301	478868	1064169
		121/04	1 0.00	501500		
	7 071•0	121694	4.285	521503	430500	952003
0-1 60	י דוד ה	10/170	4 242	541500	وسريد ورسم ر و	
		124112	4.302	541599	465137	1006736
2-2 00	) <u>610</u> 5	102575		111070	7//751	0.2.1.2.20
0-4 60		102575	4.733	404717	300291	831230
2-1 61	9865	151100	4 602	605000	251011	1050///
3-2 61		191102	4.002	092000	324804	1050664
3-3 61	705.0	127004	4 895	626110	517954	114 2044
0 - 4 61		L2 7 704	τ <b>ι</b> () ()	020110	711034	114 3704
9 - 1 62	409.5	65838	5,153	359847	313609	673456
9 - 2 67			24122	11000	515007	074210
2-3 62	396.0	66569	5,163	343663	335604	679267
Q-4 62				5,5005	333001	01 7201
Q-1 63	379.5	64663	5.696	368338	351325	719663
Q-2 63	3				****	12,000
2-3 63	418.5	7 02 4 3	6.241	438378	358111	796489
Q-4 63	i					
Q-1 64	357.0	60845	6.010	365666	365952	731618
Q-2 64	÷				,	
Q-3 64	270.0	47435	6.129	290740	298583	589323
2-4 64						
Q-1 65	102.0	17734	6.720	119172	121848	24 10 20
0 - 2 65						

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## DE SIGN / ENGINEERING 4-SYSTEM 2 5-SUBSYSTEM 22 PROGRAM MANAGEMENT

SUBD OF WORK DESIGN/ENGINEERING

LABOR + BURDEN \$	BURDEN DOLLARS	LABCR DOLLARS	LABCR RATE	L ABOR HOURS	MAN- MON TH S	
77214	38579	38635	6.602	5852	34.5	Q-3 55 Q-4 65
4919	2122	2797	8.686	322	1.5	Q-1 66
11439723	5223627	6216096		1270491	7392.0	TOTAL

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 2 PROGRAM MANAGEMENT 5-SUBSYSTEM 22 PROGRAM MANAGEMENT SUBD OF WORK DESIGN/ENGINEERING

•		MAN- Months	LABOR HOUR S	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q-1 Q-2	58 58	1.5	144	4.701	677	497	1174
Q-3 Q-4	58 58	-1.5	-144	4.701	-677	-453	-1130
Q-1 0-2	59 59		24	2.500	60	66	126
Q-3	59 59		-22	• 227	-5	-19	-24
Q - 1	60		-14	7.857	-110	-33	-143
Q-3	60 60						
Q-4 Q-1	61						
Q-2 Q-3	61						
Q-4 Q-1	61 62	18.0	3189	5.934	18922	5984	24906
Q-2 Q-3	62 62	28.5	4880	7.074	34519	4933	39452
Q-1	62 63	27.0	4713	5.989	28227	7481	35708
Q-2 Q-3	63 63	22.5	3856	6.405	24698	5498	30196
тот	AL	96.0	16626		106311	23954	130265

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM	2		
5- SUB SYSTEM	22	PROGRAM	MANAGEMENT
SUBD OF WORK	DESIGN/ENGINEEF	RING	

		MAN- MON TH S	L ABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$	EN GR MATL
Q-1	58 59	159.0	26535	4.407	116950	120575	237525	
Q = 2 Q = 3 Q = 4	58 58	463.5	77935	4.319	336618	303889	640507	
$\sqrt[2]{-1}$	59 59	790.5	135017	4.335	585361	478934	1064295	
Q=2 Q=3 Q=4	59 50	691.5	121672	4.286	521498	430481	95 1979	878
2 - 4 2 - 1 2 - 2	60 60	717.0	124158	4.361	541489	465104	1006593	-908
Q - 3	60 60	610.5	102575	4.533	464979	366251	831230	-205
Q = 4 Q = 1 Q = 2	61 41	886.5	151182	4.602	695800	354864	1050664	151
q=2 q=3	61 61	705.0	127904	4.895	626110	517854	1143964	3151
Q = 4 Q = 1 Q = 2	62 62	427.5	73027	5.187	378769	319593	698362	
Q=2 Q=3	62 62	424.5	71449	5.293	378182	340537	718719	114
Q-1 Q-2	63 63	406.5	69376	5.716	396565	358806	755371	
Q=2 Q=3 Q=4	63 63	441.0	74099	6.249	463076	363609	826685	
0 - 1	64 64	357.0	60845	6.010	365666	365952	731618	
2-2 2-3	64 64	270.0	47435	6.129	290740	298583	589323	643
Q = 4 Q = 1 Q = 2	65 45	102.0	17734	6.720	119172	121848	24 10 20	87
Q-3 Ω-4	65 65	34.5	5852	6.602	38635	38579	77214	

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM2PROGRAM MANAGEMENT5-SUBSYSTEM22SUBD OF WORKDESIGN/ENGINEERING

	MAN- Months	L AB OR H O UR S	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$	EN GR MATL
Q-1 66	1.5	322	8.686	2797	2122	4919	
TOTAL	7488.0	1287117		6322407	5247581	1156 9988	3911

APRIL 1972

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# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM	2	PROGRAM	MANAGEMENT
5-SUBSYSTEM	22		*** **** * ***************************
SUBD OF WORK	DESIGN/ENGINEER	ING	

		0 THER	SUB		T OT AL
	MPC	COST	TOTAL	G & A	COST
Q-1 58		1275	238800		238800
Q-2 58					
Q-3 58		11637	652144		652144
2-4 58	•				
2-1 59		2 3752	1088047		1088047
Q-2 59					
Q-3 59	74	35817	988748		988748
Q-4 59					
Q-1 60	-119	62913	1068479	20358	1088837
2-2 60					
Q-3 60	- 27	42649	873647	16646	890293
Q-4 60					
Q-1 61	13	56612	1107440	<b>2</b> 0580	1128020
2-2 61					
Q-3 61	266	31793	1179174	21913	1201087
Q - 4 61					
Q-1 62		2 9 8 9 4	728256	12224	740480
2-2 62	-				
Q-3 62	9	36074	754916	12671	767587
Q-4 62					
Q-1 63		22874	778245	13012	791257
2-2 63		770/	00///0	10050	<u> </u>
Q-3 63		1184	834469	13952	848421 *
2-4 63		70 /	7777744	16500	7/700/
Q=1 64		12.5	132344	15582	141926
9-2 64		707	E 00027	10574	(0250)
y = 3 64	234	121	590921	12574	003501
J-4 04	27	2202	262225	(40)	240927
Q-1 05	26	2202	243333	0472	249821
1-2 05 0-2 4F		0.01	70005	2004	00170
Q-3 05		881	18095	2084	80119
W-4 05					

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 PROGRAM MANAGEMENT 5-SUBSYSTEM 22 SUBD OF WORK DESIGN/ENGINEERING

	MPC	O THER Cost	S UB Total	G & A	T OT AL COS T
Q-1 66		63	4982	150	5132
TOTAL	476	367673	11942048	168238	12110286



#### COST DEFINITION

#### LOGISTICS SUPPORT

WBS CODE: 2.23

Total costs of \$2,717,771 contained in this WBS item reflect all identifiable in-house costs incurred in providing field service, performing maintenance analysis, producting maintenance schedules and conducting packaging studies. Included are such tasks as:

- a) Design and development of a maintenance concept
- b) Support operation analysis
- c) Development of a product performance reporting system
- d) Analysis of product performance data collected from test operations.
- e) Analysis, determination and delineation of individual tasks, expressed in terms of methods, equipment, facilities and personnel required to sustain the end product.

Detail of the recorded costs associated with this WBS item is provided by Element of Cost (EOC) and Subdivision of Work (SDW). Section III of Volume I provides a detail definition of these items.

> COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 2 5-SUBSYSTEM 23 LOGISTICS SUPPORT

			DESIGN /ENGR HOURS DOLLARS	TOTAL Hours Dollars
DESIGN/ENGINEER	ING		257696	257696
LABOR AT \$	4.554		1173517	1173517
ENGR BURDEN	AT \$	5.055	1302642	1302642
SHOP SUPPORT			18407	18407
LABOR AT \$	4.577		84244	84244
MFG BURDEN	AT \$	5.399	99385	99385
ENGR MATERIAL			6808	6808
MPC			578	578
OTHER COST			3274	3274
SUB-TOTAL			2670448	2670448
GEN & ADMIN			47323	47323
TOTAL COST			2717771	2717771

TIME-PHASED COST DETAIL - SEE PAGE II-186

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING

4-SY S	TEN	1	2	TO	TOTTOS	SIPPORT
5-SUB	SYS	TEM	23		TOTTOD	DOLLOINI
SUBD	OF	WORK	DESIGN/ENGINE	ERI	NG	

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-1	58	1.5	271	4.236	1148	1235	2383
Q-2	58						
0-3	58	4.5	688	3.865	2659	2637	5296
Q-4	58						
Q-1	59	57.0	9837	3.507	34497	37191	71688
Q-2	59						
Q-3	59	172.5	30452	3.958	120539	120123	240662
Q-4	59						
Q-1	60	84.0	14619	4.083	59695	63441	123136
Q-2	60						
0-3	60	46.5	7883	4.619	36411	32 6 2 5	69036
Q-4	60						
Q-1	61	384.0	65507	4.195	274812	260807	535619
Q-2	61						
Q-3	61	72.0	12951	4.636	60043	86406	146449
Q-4	61						
Q-1	62	60.0	10181	4.846	49336	45911	95247
0-2	62						
Q-3	62	61.5	10208	4.710	48080	51260	99340
Q-4	62						
Q-1	63	66.0	11377	4.169	47431	61776	109207
Q-2	63		,				
Q-3	63	112.5	18998 、	4.877	92650	110720	203370
Q-4	63						
Q-1	64	118.5	20249	4.862	98457	122510	220 <b>967</b>
Q-2	64						
Q-3	64	66.0	11659	5.089	59332	80191	139523
Q-4	64						
Q-1	65	72.0	12541	5.637	70696	83659	154355
0-2	65						

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## DESIGN/ENGINEERING

4-SYSTEM2LOGISTICS SUPPORT5-SUBSYSTEM23SUBD OF WORK DESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOUR S	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q-3	65	55.5	9224	5.808	53576	60032	113608
Q-4	65						
Q-1	66	39.0	6639	5.638	37432	49801	87233
Q-2	66						· •
Q-3	66	26.5	4412	6.057	26723	32317	59040
тот	AL	1499.5	257696		1173517	1302642	2476159

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 2 LOGISTICS SUPPORT 5-SUBSYSTEM 23 SUBD OF WORK DESIGN/ENGINEERING

		MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	59		40	2.150	86	111	197
Q-4	59						171
Q-1	60						
Q-2	60						
Q-3	60						
Q-4	60						
Q-1	61						
Q-2	61						
Q-3	61						
Q-4	61						
Q-1	62						
Q-2	62						
Q-3	62						
Q-4	62						
Q-1	63	85.5	14628	4.644	67927	79429	147356
Q-2	63						
Q-3	63	22.5	3739	4.341	16231	19845	36076
тот	AL	108.0	18407		84244	99385	183629

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 5-SUBSYSTEM 23 LOGISTICS SUPPORT

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		MAN- MON THS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$	ENGR MATL
Q-1	58	1.5	271	4.236	1148	1235	2383	
Q-2	58							
Q-3	58	4.5	688	3.865	2659	2637	5296	
Q-4	58							
Q-1	59	57.0	9837	3.507	34497	37 191	71688	
Q-2	59							
Q-3	59	172.5	30492	3.956	120625	120234	240859	285
Q-4	59							
Q-1	60	84.0	14619	4.083	59695	63441	123136	
Q-2	60							
Q-3	60	46.5	7883	4.619	36411	32 6 2 5	69036	
Q-4	60							
Q-1	61	384.0	65507	4.195	274812	260807	535619	6540
Q-2	61							
Q-3	61	72.0	12951	4.636	60043	86406	146449	-17
Q-4	61							
Q-1	62	60.0	10181	4.846	49336	45911	95247	
Q-2	62			-				
Q-3	62	61.5	10208	4.710	4808 C	51260	99340	
Q-4	62							
Q-1	63	151.5	26005	4.436	115358	141205	256563	
Q-2	63							
Q-3	63	135.0	22737	4.789	108881	130565	239446	
Q-4	63		,					
Q-1	64	118.5	20249	4.862	98457	122510	220 <b>967</b>	
Q-2	64							
Q-3	64	66.0	11659	5.089	59332	80191	139523	
Q-4	64							
Q-1	65	72.0	12541	5.637	70696	83659	154355	
Q-2	65							
Q-3	65	55.5	9224	5.808	53576	60032	113608	
Q-4	65							•

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 5-SUBSYSTEM 23 LDGISTICS SUPPORT

	MAN- Months	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR Matl
Q-1 66 Q-2 66	39.0	6639	5.638	37432	49801	87233	
Q-3 66	26.5	4412	6.057	26723	32317	59040	
TOTAL	1607.5	276103		1257761	1402027	2659788	6808

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 5-SUBSYSTEM 23 LOGISTICS SUPPORT

	MPC	OTHER Cost	SUB Total	GEA	TOTAL Cost
Q-1	58		2383		2383
Q-2	58 58		5296		5296
Q-4 9 Q-1 9	58 59		71688		71688
Q-2	59 59 24		241168		241168
Q-4	59 60	2085	125221	2386	127607
Q-2	60	2005	the first of first first the	2000	10,001
Q-3	60		69036	1315	70351
0-1	61 553	19	542731	10086	552817
0-2	61	5			
Q-3	61 1		146433	2721	149154
Q-4	61 62		05247	1500	24840
0-2	62		77241	1777	700-0
0-3	62		99340	1667	101007
Q-4	62				
Q-1	63		256563	4290	260853
Q-2	63	2/7	222712	(	2/2721
Q = 3	63 42	201	239113	4008	243721
0 - 1	64	,	220967	4702	225669
Q-2	64				
Q-3	64		139523	2969	142492
Q-4	64				
Q-1	65		154355	4118	158473
0-2	65 7 F	£11	114110	2015	1171/4
0-4	0) 65	211	114119		11/104
- <b>w</b> - <b>T</b> - <b>1</b>	<b>U</b> J				

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# 4-SYSTEM 2 5-SUBSYSTEM 23 LOGISTICS SUPPORT

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	MPC	OTHER COST	SUB TOTAL	GEA	TOTAL Cost
Q-1 66			87233	2627	89860
Q-3 66		392	59432	1790	61222
TOTAL	578	3274	2670448	47323	2717771


#### COST DEFINITION

#### WEAPON SYSTEM

#### WBS CODE: 2.24

Recorded cost of \$136,524,947 includes all identifiable costs incurred against the offensive and defensive weapon system. The weapon system excludes the Airborne Cooperational and the Ground Cooperational system. The original RS-70 program included both an offensive and a defensive system, however, when the program was redirected in late 1959, the requirement for the weapon system was deleted. With the addition of AV No. 3 to the B-70 program, the weapon system was reinstated.

A summary of the subcontractor recorded cost data is presented on page II-195 The vast majority of the costs accumulated in this WBS item are associated with IBM's development activities on the Bomb/Navigation System. A thorough discussion of the contractual arrangements, program redirections, delivery dates and other pertinent supplier data is provided. Cost data includes the supplier expenditures for engineering production, tooling and testing (where identified) performed at the supplier's facility. Refer to the Subcontracting Element of Cost definition (Volume I, page I-26) for additional explanation.

As an aid in the definition and evaluation of the in-house engineering costs associated with the weapon system, a matrix of engineering hours has been developed. This matrix recaps the hours charged by the engineering groups in support of the weapon system development. The matrix is as follows:

Group	Title	Hours Expended
3	Electrical and Avionics Installation	2,067
11	Weight Control	3,622
13	Aerodynamics	2,577
14	Wind Tunnel Models	1.678
32	Armament	14,163
35	Fuselage	2.437
45	Design Illustrations	3.875
48	Communication and Indicating System	35.654
49	Avionics Integration and Control	157.789
50	Metallurgy	5.696
55	Flight Controls Analysis	3.765
57	Engineering Specifications	17.678
61	Operations Administration	3.728
63	Flight Test Maintenance	8.768
72	Engineering Computing	2,955
75	Non-Metallics	6,587
76	Flight Test Project Engineering	8,877



WBS CODE: 2.24

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Group	Title	Hours Expended
<b>7</b> 7	Reliability and Maintainability	5,183
84	Weapon System Analysis	182,913
85	Communication and Indicating System	72,750
86	Electronic Integration	20,157
94	Flight Simulation	9,397
95	Electrical System Design	13,104
96	Wind Tunnel Projects	10,015
97	Laboratory Services	8,645
110	Electrical Power Lab	35,802
119	Preliminary Design	61,611
120	R&D Programming	13,819
125	Electrical System Equipment	6,992
127	B-70 Project Group	31,831
	Various	6,563
	Total Engineering Hours	760,698
	2.24 Engineering SDW 737,178	

2.24 Test/QC SDW 23,520

Total Hours 760,698

II-194

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WBS Code: 2.24.1

1959

#### SUBCONTRACTOR MATRIX

SUBCONTRACTOR	ENGINEERING	PROD	TOOLING	TEST	TOTAL
IBM Beech Westinghouse Others	33,252,681 1,661,291 946,753 462,603	81,183,752 862,245 2,710,735 1,097,308	12,758	-	114,436,433 2,536,294 3,657,488 1,559,911
TOTAL	36,323,328	85,854,040	12,758	-	122,190,126

#### Subsystem: Weapon Subsystem

IBM was selected to produce the AN/ASQ028(V) Bombing Navigation and Missile Guidance Subsystem for the B-70 Weapon System.

The two letter contracts awarded to IBM for this effort, along with their award and completion dates are as follows:

L861-GX-600013	May 1, 1958	-	December	2, 19
L1E1-YJ-600221	October 15, 1960	-	March 6,	1964

The Statement of Work for the two contracts required IBM to provide engineering, management, fabrication and other necessary services leading to the design, development and support of the AN/ASQ-28(V) Bombing Navigation and Missile Guidance Subsystem for the B-70. In addition, Letter Contract 600221 called for IBM to deliver one prototype model and nine flight test models. On April 3, 1961, all but the prototype model were cancelled.

Work on the Bombing Navigation and Missile Subsystem was initiated by the Air Force under Prime Contract AF33(600)-31315 to IBM. When NR was selected as the B-70 Weapon System prime contractor, the IBM prime contract was terminated and replaced by NR Purchase Order L861-GX-600013.

The AN/ASQ(V) Bombing Navigation and Missile Guidance Subsystem required the highest degree of engineering technology to develop, fabricate, and test the highly accurate system. It represented over five years of navigation utilizing digital computation with stellar-monitored, Doppler-damped inertial guidance. This, combined with high resolution, side-looking and squint capabilities as provided through the Doppler data processor, made up the most effective bombing navigation, missile guidance system ever developed for the Air Force.

On December 2, 1959, Contract 600013 was terminated for the convenience of the Government. At termination, IBM had basically completed the experimental model Bombing Navigation and Missile Guidance Subsystem program. Complete design release for fabrication was scheduled to be complete in May 1960. At the time of termination the program was well into the hardware stage as evidenced by the vast amount of inventory that had been acquired and which had to be inventoried and placed in warehouse storage for disposition. At termination, Contract 600013 was approximately 89% completed.

SD72-SH-0003



## WBS CODE: 2.24.1

On March 3, 1960 NR directed IBM to suspend all activity relative to screening of excess B-70 inventory and the Air Force issued Prime Contract AF33(600)-41253 to IBM to continue a portion of the BNMGS Program, known as the C-121 Program.

When the B-70 program was initiated in October 1960, IBM's prime contract was again terminated and NR issued Letter Contract 600221 to IBM to continue the C-121 prototype program concurrently with design, development and delivery of 12 flight test models. On April 3, 1961, effort on the flight test models was terminated but effort was continued on the prototype BNMGS, which was to be used on Air Vehicle 3. On March 6, 1964 the program was completely terminated for the convenience of the Government. IBM and its subcontractors disposed of all residual inventory in accordance with direction from the Air Force. Approximately 30,000 line items of inventory had been counted and categorized as defined by ASPR; part number, description, quantity, condition code, unit measure and cost determined and listed on worksheets. These worksheets were then verified by IBM Property Administration and Quality Control. The inventory was moved to warehouse staging area, worksheets verified against physical property and location annotated. The proceeds obtained from the disposition of the property was credited to Purchase Order 600221.

<u>BEECH</u> was selected to produce the Alert Pod Subsystem for the B-70 in accordance with Specification NA5-3093.

The two letter contracts awarded to Beech for this effort, along with their award and completion dates are as follows:

L961-GX-600124	April 6,	, 1959	-	December 3, 1959
L001-YZ-600227	November 11,	, 1960	-	March 31, 1961

The Statement of Work called for the subcontractor to perform engineering, management, and manufacturing services, including but not limited to, analytical and design studies directed toward the design, development, testing and mockup of the Alert Pod Subsystem.

Beech was in the early stages of design and development on Contract 600124 and had completed 9.7% of the effort, when it was terminated on December 3, 1961.

Letter Contract 600227 was awarded to Beech on November 11, 1960 as a continuation of the effort initiated by Purchase Order 600124. Beech was again in the early stages of design and development when the contract was terminated on March 31, 1961 for the convenience of the Government, having completed 4.5% of the effort.

<u>WESTINGHOUSE</u> was selected to provide the Defensive Subsystem Group for the B-70 Air Vehicles. Two Letter Contracts were awarded to Westinghouse for this effort:

L961-GX-600127	June 3, 1959	) (	December 2, 1959
L1E1-YZ-600320	October 29, 1960	) –	March 31, 1961

The Statement of Work for the two contracts called for the subcontractor to furnish analytical, design, test, and other necessary studies and programs as



WBS CODE: 2.24.1 required by NR's governing specification, leading to the definition and optimization of the Defensive Subsystem Group.

Westinghouse was in the early stages of design and development when Contract 600127 was terminated on December 2, 1959, for the convenience of the Government.

Letter Contract 600320 was awarded to Westinghouse on October 29, 1960 as a continuation of the effort initiated by Purchase Order 600127. Under the terms of the new purchase order all terminated inventory generated against Letter Contract 600127 and not disposed of previously was to be reviewed for usability and transferred to Contract 600320 at no cost. In addition, the subcontractor was required to conduct a complete investigation and evaluation of the AN/ALQ-27 Defensive Subsystem Program which was being developed under an Air Force Prime Contract placed with Sperry Gyroscope Company. The study objectives were:

- 1. AN/ALQ-27 program areas which were applicable to a YB-70 Defensive Subsystem.
- 2. AN/ALQ-27 program areas which were not applicable to a YB-70 Defensive Subsystem and reasons.
- 3. Specific recommendations for modification or expansion of the AN/ALQ-27 Flight Test Program which were necessary to orient or to assist in the design of a YB-70 Defensive Subsystem.
- 4. Technical assistance and study effort as may be necessary to define a YB-70 Defensive Subsystem.

The AN/ALQ-27 study effort was completed on January 16, 1961 and the design and development effort called for under Contract 600320 was in the early stages of design on March 31, 1961, the date the contract was terminated.

Three companies submitted proposals in competition for the B-70 Defensive Subsystem. Westinghouse Air Arm Division, Baltimore, Maryland, was selected as Defensive Subsystem subcontractor in April of 1959, primarily on the basis of their advanced engineering concept. This concept was considered a prime factor, since utilization of the Defensive Subsystem for the B-70 required a combination of techniques beyond the capability of any single company. It was North American Rockwell's and Westinghouse's intention to utilize the countermeasures industry to the fullest extent on this program.

The various tier subcontractors were to be selected from those industry members whose technical capability was proven. To implement this approach, a detailed technical development survey was completed on November 1959. This was to aid in the selection of lower tier contractors in December 1959. However, prior to this the Defensive Subsystem contract with Westinghouse was terminated. All inventory and tooling related to the above contracts were disposed of in accordance with Air Force direction and the cost credited to the appropriate contract.

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# COST BREAKDOWNS B-70 AIRCRAFT STUDY

# 4-SYSTEM 2 5-SUBSYSTEM 24 WEAPON SYSTEM

		DESIGN /ENGR HOURS DOLLARS	PROD Hours Dollars	TOOLING AND STE HOURS DOLLARS	TEST ZQC HOURS DOLLARS
DESIGN/ENGINEERING		737178			23520
LABOR AT \$ 4.614		3411101			98979
ENGR BURDEN AT \$	4.201	3095834			100135
PRODUCT ION			220672		
LABOR AT \$ 3.646			804564		
SHOP SUPPORT		1406			13004
LABOR AT \$ 3.047		4493			39408
PLANN ING			48		53
LABOR AT \$ 4.495			163		291
TEST/QC		131	1084		288
LABOR AT \$ 3.627		423	4216		813
MFG BURDEN AT \$	1.729	5780	42262		361114
ENGR MATERIAL		1175			49176
MFG MATERIAL			5575		
SUBCONTRACT		36323328	85854040	12758	
MPC		1058974	3544765	589	4079
OTHER COST		287880			113
SUB-TOTAL		44188988	90255585	13347	654108
GEN & ADMIN	~	269601	1131495	174	11649
TOTAL COST		44458589	91387080	13521	665757

TIME-PHASED COST				,
DETAIL - SEE PAGE	II <b>-</b> 200	II-208	II-213	II-214

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# COST BREAKDOWNS B-70 AIRCRAFT STUDY

TOTAL

4-SYSTEM 2 5-SUBSYSTEM 24 WEAPON SYSTEM

	HOURS
	DOLLARS
DESIGN/ENGINEERING	760698
LABOR AT \$ 4.614	3510080
ENGR BURDEN AT \$	4.201 3195969
PRODUCTION	220672
LABOR AT \$ 3.646	804564
SHOP SUPPORT	14410
LABOR AT \$ 3.047	43901
PLANN ING	101
LABOR AT \$ 4.495	454
TEST/QC	1503
LABOR AT \$ 3.627	5452
MFG BURDEN AT \$	1.729 409156
ENGR MATERIAL	50351
MFG MATERIAL	. 5575
SUBCONTRACT	122190126
MPC	4608407
OTHER COST	287993
	مجرد الألب مريح وغنيه فينه جمل مريح شاله
SUB-TOTAL	135112028
GEN & ADMIN	1412919
TOTAL COST	136524947

TIME-PHASED COST, DETAIL - SEE PAGE

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING 4-SYSTEM 2 5-SUBSYSTEM 24 WEAPON SYSTEM SUBD OF WORK DESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOUR S	LABOR	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58	299.0	50268	4.564	229446	228716	458162
Q-2	58						
Q-3	58	580.5	97503	4.252	414 55 8	384293	798851
Q-4	58						
Q-1	59	502.5	85848	4.171	358034	296489	654523
Q-2	59						
Q-3	59	621.0	109344	4.037	441428	392755	834183
Q-4	59						
Q-1	60	363.0	62845	4.309	270803	233064	503867
Q-2	60						
0-3	60	145.5	24426	4.875	119080	89465	208545
0 - 4	60						
Q-1	61	451.5	77084	4.766	367344	257976	625320
Q-2	61	1/0					
Q-3	01	162.0	29441	5.032	148167	148914	297081
Q-4	01	204 0	2/720	5 307	1070/7	1 = 2 / 2 0	
0-2	02 40	204.0	34129	2.300	18/06/	159492	340009
0-2	02 62	207 0	21726	6 067	160276	175540	272014
0-4	62	20100	54150	4041	100514	115540	242714
0 - 1	63	214.5	36617	5.490	201033	108101	200124
0-2	63		50011	24430	201000	1 90 101	222124
0-3	63	280.5	47176	5,287	249420	272203	521623
0-4	63					The E Carl Star Co rate	> L + VL -
Q-1	64	250.5	42674	5.397	230309	231631	461940
Q-2	64						
Q-3	64	24.0	4254	5.840	24845	25696	50541
Q-4	64						
Q-1	65	1.5	159	5.252	835	1050	1885
Q-2	65			·			

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

DESIGN/ENGINEERING 4-SYSTEM 2 5-SUBSYSTEM 24 WEAPON SYSTEM SUBD OF WORK DESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOURS	LABOR Rate	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 65		63	5.302	334	419	753
Q-4 65 Q-1 66		5	4.800	24	30	54
TOTAL	4307.0	737178		3411101	3095834	6506935

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP	SUPPORT	
4-SYSTEM	2	WEADON	CVCMEM
5-SUBSYSTEM	24		OTOTEM
SUBD OF WORK	DESIG	N/ENGINEERING	

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-1	58 50	1.5	146	2.575	376	484	86 0
Q-2	28						
Q-3	58		80	6.800	544	313	857
Q-4	50						
0-1	59 50		53	2.774	147	185	332
Q-2	59	( )					
0 4	29 50	6.0	995	2.934	2919	4049	6968
0-1	27	, r					
Q-1	60		-676	2.883	-1949	-1833	-3782
Q-2	6U 4 0						
0-4	60 40					-59	-59
N-1	60 41	. Ε	201	2 2 4 2			
0-2	01 41	4.0	804	3.041	2445	2384	4829
0-2	61 61						
0-4	61					240	240
0-1	62		1. 1.	2 077	1 2 1	200	
0-2	52		***	2.911	101	200	331
0-3	62		-41	3 000	-122	- 140	201
0-4	62		7 T	000 •C	-125	-100	-291
0-1	63						
0-2	63		,				
Q-3	63		1 🔍	3.000	3	-15	-12
TOT	AL	7.5	1406		4493	5780	10273

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 2 WEAPON SYSTEM 5-SUBSYSTEM 24 SUBD OF WORK DESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	59 50		35	2.829	99		99
Q = 1 Q = 2	60 60		3	5.667	17		17
Q-3 0-4	60 60						•
Q - 1 Q - 2	61 61	•	92	3.304	304		304
Q-3 Q-4	61 61		1	3.000	3		3
Q-1 Q-2	62 62						
Q-3 Q-4	62 62		5	7.200	36		36
Q-1 Q-2	63 63				•		
Q-3	63		-5	7.200	-36		-36
тот	AL		131		423		423

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 5-SUBSYSTEM 24 SUBD OF WORK DESIGN/ENGINEERING

		MAN- MON TH S	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR Matl
Q-1	58 59	300.5	50414	4.559	229822	229200	459022	458
Q-3	58 58	580.5	97583	4.254	415102	384606	799708	-33
Q = 4 Q = 1 Q = 2	50 59	502.5	85901	4.170	358181	296674	654855	-77
Q-3 0-4	59 59	627.0	.110374	4.027	444446	396804	841250	789
0 - 1 0 - 2	60 60	358.5	62172	4.325	268871	231231	500102	-654
Q-3 0-4	60 60	145.5	24426	4.875	119080	89406	208486	38
Q-1 Q-2	61 61	456.0	77980	4.746	370093	260360	630453	560
Q-3	61 61	162.0	29448	5.032	148170	149154	297324	-28
0 - 1 0 - 2	62 62	204.0	34773	5,383	187198	159692	346890	129
Q-3 Q-4	62 62	207.0	34700	4.850	168287	175372	343659	-7
Q-1 Q-2	63 63	214.5	36617	5.490	201033	198101	399134	
Q-3 Q-4	63 63	280.5	47172	5.287	249387	272188	521575	
Q-1 Q-2	64 64	250.5	42674	5.397	230309	231631	<b>4619</b> 40	
Q-3 Q-4	64 64	24.0	4254	5.840	24845	25696	50 <b>541</b>	-1
Q-1 Q-2	65 65	1.5	159	5.252	835	1050	1885	1
Q-3 Q-4	65 65		63	5.302	334	419	753	

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM2WEAPON SYSTEM5-SUBSYSTEM24SUBD OF WORKDESIGN/ENGINEERING

	MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$	ENGR Matl
Q-1 66		5	4.800	24	30	54	
TOTAL	4314.5	738715		3416017	3101614	6517631	1175

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM2WEAPON SYSTEM5-SUBSYSTEM24WEAPON SYSTEMSUBD OF WORKDESIGN/ENGINEERING

		SUBC	TOTAL MATERIAL	MPC	OTHE R COS T	SUB Total	GεA	TOTAL Cost
Q-1	58		458	25	47	459552		459552
0-2	20	7447000	7//70/7	17000	<b></b>			
0-4	50	1001900	1001801	67998	5861	8541434		8541434
0-1	50	15272270	16272200	407/7/	100007			
n - 2	50	10010019	12212202	40/034	100807	16536598		16536598
0-3	59	2813500	2816280	76 04 5	20210	27/1702		
0-4	59	2013300	2014203	10 94 5	29218	3101102		3761702
$\tilde{0}-1$	60	1000807	1000153	50201	11575	1671101	20025	1/0107/
$\bar{0} - 2$	60	******	1000100	27271	11212	12/11/21	29935	1001056
0-3	60	1289784	1289822	76531	11221	1596060	20.21.0	1/1/070
Q-4	60			10001		1200000	50219	1010/19
Q-1	61	1005368	1005928	28933	30895	1696209	21521	1727720
Q-2	61			20100	50075	10 /0 20 /	51 72 1	1121150
Q-3	61	1140320	1140292	32669	14810	1485095	27598	1512603
Q-4	61				21010	1.00070	21000	1712073
Q-1	62	1347123	1347252	42 82 4	-10761	1726205	28974	1755179
Q-2	62							1100110
Q-3	62	1347122	1347115	42775	33208	1766757	29655	1796412
Q-4	62							1130124
Q-1	63	2080310	2080310	88388	33269	2601101	43490	2644591
0-2	63							
Q-3	63	359054	359054	11539	5435	897603	15008	912611
Q-4	63							
0-1	64	898661	898661	123422	20816	1504839	32020	1536859
Q-2	64							
Q-3	64		-1	·	1099	51639	1099	52738
Q-4	64							
Q-1	65		1		266	2152	57	2209
<u>u-2</u>	65							
<b>₩</b> -3	05				106	859	23	882
V-4	やり							

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM2WEAPON SYSTEM5-SUBSYSTEM24SUBD OF WORKDESIGN/ENGINEERING

TOTAL COST	G & A	SUB Total	OTHER COST	MPC	TOTAL MATERIAL	SUBC	
64	2	62	8				Q-1 66
44458589	269601	44188988	287880	1058974	36324503	36323328	TOTAL

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	PRODUCTION
4-SYSTEM	2
5-SUB SYSTEM	24
SUBD OF WORK	PRODUCTION

TOTAL 1264.5

WEAPON SYSTEM

804564

42194

846758

#### **ON-SITE LABOR**

220672

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	59 50	576.0	101470	3.689	374296		374296
Q-1 Q-2	60 60	625.5	108474	3.643	395211		395211
Q-3 Q-4	60 60						
Q-1 Q-2	61 61	63.0	10648	3.239	34484	39347	73831
Q-3 Q-4	61 61		80	7.163	573	2847	3420
Q-1	62						
Q-3	62 62						
Q - 4	62 63						
0-2	63						
Q-3	63				•		

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

WEAPON SYSTEM

## PLANNING 4-SYSTEM 2 5-SUBSYSTEM 24 SUBD OF WORK PRODUCTION

	MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1 61		48	3.396	163	68	231
TOTAL		48		163	68	231

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# TEST/QC

4-SYSTEM 2 5-SUBSYSTEM 24 SUBD OF WORK PRODUCTION

WEAPON SYSTEM

		MAN- MONTHS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1 Q-2 Q-3	60 60 60		112	7.241	811		811
Q-4 Q-1	60 61	6.0	972	3.503	3405		3405
TOT	AL	6.0	1084		4216		4216

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 5-SUBSYSTEM 24 SUBD OF WORK PRODUCTION

WEAPON SYSTEM

		MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$	MF( Matl
Q-3	59	576.0	101470	3.689	374296		374296	
Q-4	59							-
Q-1	60	625.5	108586	3.647	396022		396022	
Q-2	60							
Q-3	60							
Q-4	60							
0-1	61	69.0	11668	3.261	38052	39415	77467	4780
0-2	61							
0-3	61		80	7.163	573	2847	3420	795
0-4	61					2011	0 120	
0-1	62							
0-2	62							
0-3	62							
0-4	62							
0-1	63		•	,				
$\bar{Q}-2$	63							
Q-3	63							
тот	AL	1270.5	221804		808943	42262	851205	5575

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTE	M	2		
5-SUB SY	STEM	24	WEAPON	SYSTEM
SUBD OF	WORK	PRODUCTION		

			TOTAL		SUB		TOTAL
		SUBC	MATERIAL	MPC	TOTAL	G&A	COST
Q-3	58	403574	403574	3581	407155		407155
Q-4	58						
Q-1	59	3050417	3050417	80884	3131301		3131301
Q-2	59						
Q-3	59	24671259	24671259	673913	25719468		25719468
Q-4	59						
Q-1	60	18527001	18527001	1099276	20022299	381485	20403784
Q-2	60						
Q-3	60	8625848	8625848	511770	9137618	174099	9311717
Q4	60						
Q-1	61	8648685	8653465	2 <b>4</b> 819 <b>7</b>	8979129	166859	9145988
Q-2	61						
Q-3	61	8536107	8536902	244778	8785100	163254	8948354
Q 4	61						
0-1	62	2991441	2991441	95 07 3	3086514	51807	3138321
0-2	62						
Q3	62	2991433	2991433	94986	3086419	51805	3138224
Q-4	62						
Q-1	63	4795488	4795488	203635	4999123	83585	5082 <b>70</b> 8
Q~ L	63	e					
Q~3	63	666689	666689	21463	688152	11506	699658
Q=4	03		******				
N T	64	1946098	1946098	267209	2213307	47095	2260402
TOT	AL	85854040	85859615	3544765	90255585	1131495	91387080

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM2WEAPON SYSTEM5-SUBSYSTEM24SUBD OF WORK TOOLING AND STE

			TOTAL		SUB		TOTAL	
		SUBC	MATERIAL	MPC	TOTAL	G&A	COST	
Q-3	59	4035	4035	110	4145		4145	
Q-4	59							
Q-1	60	4278	4278	253	4531	86	4617	
Q-2	60							
Q-3	60	3286	3286	194	3480	66	3546	
Q-4	60							
Q-1	61	1142	1142	32	1174	22	1196	
Q-2	61							
Q-3	61	17	17		17		17	
тот	AL	12758	12758	589	13347	174	13521	

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESIGN/ENGINEER	ING	
4-SYSTEM	2		
5-SUB SYSTEM	24	WEAPON	SYSTEM
SUBD OF WORK	TEST/QC		

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-1	<b>6</b> 0						
Q-2	60						
0-3	60		1	3.000	3		3
Q-4	60						
Q-1	61	76.5	13034	4.400	57352	44305	101657
Q-2	61						
Q-3	61	5.5	951	3.843	3655	10417	14072
Q-4	61						
Q-1	62	28.5	4949	3.978	19687	21224	40911
Q-2	62						
Q-3	62	16.5	2888	3.604	10409	13638	24047
Q-4	62						
Q-1	63	11.5	1919	4.498	8632	10422	19054
Q-2	63						
Q-3	63	-1.5	-223	3.399	-758	128	-630
Q-4	63						
Q-1	64		1	• 99 9	-1	1	
тот	TAL	137.0	23520		98 <b>97</b> 9	100135	199114

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP SUPPORT	
4-SYSTEM	2	
5-SUB SYSTEM	24	WEAPON SYSTEM
SUBD OF WORK	TEST/QC	

		MAN- Months	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q-3	58	4.5	669	2.709	1812	2588	4400
Q-4	58						
Q-1	59	4.5	886	2.837	2514	3359	5873
Q-2	59						
Q-3	59	6.0	993	2.763	2744	4122	6866
Q-4	59						
Q-1	60	10.5	1759	2.895	5093	12203	17296
Q-2	60						
Q-3	60	6.0	914	2.745	2509	187088	189597
Q-4	60						
Q-1	61	35.0	5998	3.167	18996	<b>99759</b>	118755
Q-2	61						
Q-3	61	2.5	411	3.416	1404	46572	47976
Q-4	61						
Q-1	62		122	2.533	309	443	752
Q-2	62						
Q-3	62	4.5	677	3.065	2075	2685	4760
Q-4	62						
Q-1	63	1.5	176	3.295	580	715	1295
Q-2	63						
Q-3	63	3.0	401	3.426	1374	1581	2955
Q-4	63						
Q-1	64		-2	. 1.000	-2	-1	-3
тот	AL	78.0	13004		39408	361114	400522

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PLANNING 4-SYSTEM 2 5-SUBSYSTEM 24 SUBD OF WORK TEST/QC

WEAPON SYSTEM

	MAN- MON THS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1 63 Q-2 63		28	3.357	94	111	205
Q-3 63		25	7.880	197	-111	86
TOTAL		53		291		291

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 2 5-SUBSYSTEM 24 SUBD OF WORK TEST/QC

WEAPON SYSTEM

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	58		44	2.659	117		117
Q-4	58		24		~ ~		
Q-1	27 50		24	3.333	80		80
0-2	50		20	2 044	07		
0-4	55		29	2.900	80		80
0-1	60		4	2 750	11		11
0-2	60		-	2.150	11		11
0-3	60						
0-4	60						
0-1	61	1.0	135	2.741	370		370
Q-2	61				0.0		5.0
Q-3	61		16	2.500	40		40
Q-4	61						
Q-1	62		-3	2.000	-6		-6
Q-2	62						
Q-3	62		28	2.786	78		78
Q-4	62	•					
Q-1	63						
Q-2	63						
Q-3	63		9	3.889	35		35
Q-4	63				_	•	
Q-1	64		2	. 1.000	2		2
TOI	<b>AL</b>	1.0	288		813		813

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

WEAPON SYSTEM

4-SYSTEM 2 5-SUBSYSTEM 24 SUBD OF WORK TEST/QC

		MAN- MONTHS	LABOR HOUR S	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR Matl
Q-3 Q-4	58 58	4.5	713	2.705	1929	2588	4517	2199
Q-1 Q-2	59 59	4.5	910	2.851	2594	3359	5953	590
Q-3 Q-4	59 59	6.0	1022	2.769	2830	4122	6952	239
0-1 Q-2	60 60	10.5	1763	2.895	5104	12203	17307	-900
Q-3 Q-4	60 60	6.0	915	2.745	2512	187088	189600	146
Q-1 Q-2	61 61	112.5	19167	4.003	76718	144064	220782	12864
Q-3 Q-4	61 61	8.0	1378	3.700	5099	56989	62088	8835
Q-1 Q-2	62 62	28.5	5068	3.944	19990	21667	41657	20574
Q-3 Q-4	62 62	21.0	3593	3.496	12562	16323	28885	-1191
Q-1 Q-2	63 63	13.0	2123	4.383	9306	11248	20554	6109
Q-3 Q-4	63 63	1.5	212	4.000	848	1598	2446	-281
Q-1	64	_	1	• 99 9	-1		-1	-8
TOT	AL	216.0	36865		139491	461249	600740	49176

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

WEAPON SYSTEM

4-SYSTEM 2 5-SUBSYSTEM 24 SUBD OF WORK TEST/QC

OTHER SUB TOTAL MPC COST G & A TOTAL COST Q-3 58 120 6836 6836 Q-4 58 0-1 59 50 6593 6593 Q-2 59 Q-3 59 20 7211 7211 Q-4 59 Q-1 60 -118 16289. 310 16599 Q-2 60 Q-3 60 19 189765 3616 193381 Q-4 60 Q-1 61 1087 234733 4362 239095 Q-2 61 Q-3 61 746 113 71782 1334 73116 Q-4 61 Q-1 62 1621 3 1072 63855 64927 Q-2 62 Q-3 62 -93 19 27620 464 28084 Q-4 62 Q-1 63 601 27264 455 27719 0-2 63 Q-3 63 27 -22 2170 36 2206 Q - 4 63Q-1 64 -1 -10 -10 113 TOTAL 4079 654108 11649 665757

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING

4-SYSTEM 2 5-SUBSYSTEM 24 WEAPON SYSTEM

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58	299.0	50268	4.564	229446	228716	458162
Q-2	58						
Q-3	58	580.5	97503	4.252	414558	384293	798851
Q-4	58						
Q-1	59	502.5	85848	4.171	358034	296489	654523
Q-2	59						
Q=3	59	621.0	109344	4.037	441428	392755	834183
0 - 1	59 60	363 0	67915	4 200	370003	2220/1	E 0 3 0 ( 7
$\overline{Q}$ - 2	60		02040	4.303	210005	2 3 3 0 0 4	203867
Q-3	60	145.5	24427	4.875	119083	89465	208548
Q-4	60						
Q-1	61	528.0	90118	4.713	424696	302281	726977
0-2	61						
Q-3	61	168.0	30398	4.994	151822	159331	311153
Q-4	61						
Q-1	62	232.5	39678	5.211	206754	180716	387470
<u>u-</u> 2	62						
Q = 3	62	223.5	37624	4.752	178783	189178	367961
0 1	62	22 C E	30504	<b>F</b> ( ) <b>S</b>	<b></b>		
Q = 1 Q = 2	63	220.0	38536	5.441	209665	208523	418188
$\tilde{0} - 3$	63	279.0	46953	5,296	248662	272331	520093
Q-4	63			20270	213002	212331	120115
Q-1	64	250.5	42675	5.397	230308	231632	461940
Q-2	64						
Q-3	64	24.0	4254	5.840	24845	25696	50541
Q-4	64						
Q <b>-1</b>	65	1.5	159	5.252	835	1050	1885
Q-2	65						

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

DESIGN/ENGINEERING 4-SYSTEM 2 5-SUBSYSTEM 24 WEAPON SYSTEM

LABOR + BURDEN \$	BUR DEN Doll Ars	LABOR DOLLARS	LABOR RATE	LABOR HOUR S	MAN- MON THS		
753	419	334	5.302	63		2-3 65	Q-3 0-4
54	30	24	4.800	5		Q-1 66	Q-1
6706049	3195969	3510080		760698	4445.0	TOTAL	TO

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# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### PRODUCTION 4-SYSTEM 2 . 5-SUBSYSTEM 24 WEAPON SYSTEM

#### ON-SITE LABOR

		MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL AR S	LABOR + BURDEN \$
Q-3 Q-4	59 59	576.0	101470	3.689	374296		374296
Q-1 Q-2	60 60	625.5	108474	3.643	395211		395211
Q-3 Q-4	60 60						
Q-1 Q-2	61 61	63.0	10648	3.239	34484	39347	73831
Q-3 Q-4	61 61		80	7.163	573	2 847	3420
Q=1 Q=2	62 62						
Q - 4 Q - 1	62 63						
Q-2 Q-3	63 63						
тот	AL	1264.5	220672		804564	42194	846758

804564 42194 846758

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 2 5-SUBSYSTEM 24 WEAPON SYSTEM

	MAN- Months	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1 58	1.5	146	2.575	376	484	860
Q-2 58 Q-3 58	4.5	749	3.146	2356	2 90 1	5257
Q-4 58						
Q-1 59	6.0	939	2.834	2661	3 5 4 4	6205
Q = 2 59	11.5	1099	2 940	5667	0171	12024
Q-4 59	1145	1,200	2 • 0 <del>4</del> 7	2005	01/1	12024
Q-1 60	6.0	1083	2.903	3144	10370	13514
Q-2 60	<i>.</i>					
Q = 3 60	6.0	914	2.745	2509	187029	189538
Q - 1 61	40.0	6802	3.152	21441	102143	123584
Q-2 61						
Q-3 61	2.5	411	3.416	1404	46812	48216
0-1 61	15	166	2 451		44.7	1000
Q-2 62	10,2	100	2.071		C+0	1003
Q-3 62	4.0	636	3.069	1952	2517	4469
Q-4 62						
Q-1 63	1.5	176	3.295	580	715	1295
$Q = 2 \ 63$	3.0	402	3.425	1377	1 566	2943
Q-4 63						
Q-1 64		-2	1.000	-2	-1	-3
TOTAL	88.0	14410		43901	366 894	410795

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# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### PLANNING 2 5-SUB SYSTEM 24

WEAPON SYSTEM

4-SYSTEM

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-1 Q-2 Q-3 Q-4 Q-1 Q-2 Q-3 Q-4	61 61 61 62 62 62 62 62		48	3.396	163	68	231
Q = 1 Q = 2	63 63		28	3.357	94	111	205
Q-3	63		2.5	7.880	197	-111	86
тот	AL		101		454	68	522

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 2 5-SUBSYSTEM 24 WEAPON SYSTEM

	MAN- MONTHS	LABOR HOUR S	LABOR	LABOR DOLLARS	BUR DEN	LABOR + BURDEN \$
Q-3 58		44	2.659	117		117
Q-4 58 Q-1 59		24	3.333	80		80
Q = 2 59 Q = 3 59 Q = 4 59		64	2.891	185		185
Q = 1 60		119	7.050	839		839
Q-3 60 Q-4 60						
Q = 1  61 Q = 2  61	7.5	1199	3.402	4079		4079
Q-3 61 Q-4 61		17	2.529	43		43
Q = 1 62 Q = 2 62		-3	2.000	-6		-6
Q-3 62 Q-4 62		33	3.455	114		114
Q = 1  63 Q = 2  63						
Q-3 63 Q-4 63		4	• 249	-1		-1
Q-1 64		2	1.000	2		2
TOTAL	7.5	1503		5452		5452

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 5-SUBSYSTEM 24 WEAPON SYSTEM

		MAN- MON THS	LABOR	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$	ENGR MATL
Q-1 Q-2	58 58	300.5	50414	4.559	229822	229200	459022	458
Q-3 Q-4	58 58	585.0	98296	4.243	417031	387194	804225	2166
Q-1 Q-2	59 59	508.5	86811	4.156	360775	300033	660808	513
Q-3 Q-4	59 59	1208.5	212866	3.860	821572	400926	1222498	1028
Q-1 Q-2	60 60	994.5	172521	3.884	66999 <b>7</b>	243434	913431	-1554 (
Q-3 Q-4	60 60	151.5	25341	4.798	121592	276494	398086	184
Q-1 Q-2	61 61	638.5	108815	4.456	484863	443 839	928702	13424
Q-3 Q-4	61 61	170.5	30906	4.978	153842	208990	362832	8807
Q-1 Q-2	62 62	234.0	39841	5.200	207188	181359	388547	20703
Q-3 Q-4	62 62	227.5	38293	4.723	180849	191695	372544	-1198
Q-1 Q-2	63 63	228.0	38740	5.430	210339	209349	419638	6109
Q-3 Q-4	63 63	282.0	47384	5.281	250235	273786	524021	-281
Q-1 Q-2	64 64	250.5	42675	5.397	230308	231631	461939	-8
Q-3 Q-4	64 64	24.0	4254	5.840	24845	25696	50541	-1
Q-2	65 65	1.5	159	5.252	835	1050	1885	1
Q−3 Q−4	65 65		63	5.302	334	419	753	:

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 5-SUBSYSTEM 24 WEAPON SYSTEM

	MAN- MON THS	LABOR	LABOR	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGF MATL
Q-1 66		5	4.800	24	30	54	
TOTAL	5805.0	997384		4364451	3605125	7969576	50351

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# 4-SYSTEM 2 5-SUBSYSTEM 24 WEAPON SYSTEM

		MFG MATL	TOOL/STE MATL	SUBC	TOTAL MATERIAL	MPC	OT HE R C OS T	SUB Total	
Q-1	58				458	25	47	459552	
0-3	20 59			2071/7/					
0-4	58			8071474	8073640	71699	5861	8955425	
$\tilde{0}-1$	59			1 94 2 3 794	19424200	100510	100007	10/7//00	
Q-2	59			10423130	10424309	400,000	100807	19674492	
Q-3	59			27488794	27489822	750988	20210	20402524	
Q-4	59					120200	27210	23432320	
Q-1	60			19532086	19530532	1158702	11575	21614240	(internet internet in
Q-2	60								
Q-3	60			9918918	9919102	588514	11221	10916923	
Q-4	60	( 700							
	01 41	4780		9655195	9673399	278249	30895	10911245	
0-3	61	705		017(11)	0101011	070100			
0-4	61	175		90/0444	9686046	278193	14923	10341994	
0-1	62			4338564	4259267	120510	10750	107/67/	
Q-2	62			1000004	+JJJ201	134310	-10758	48/62/4	
Q-3	62			4338555	4337357	137668	33227	4980706	
Q-4	62					137030	55221	4000190	
Q-1	63			6875798	6881907	292624	33269	7627488	
Q-2	63								
Q-3	63			1025743	1025462	33 0 2 9	5413	1587925	
Q-4	63								
Q - I	64			2844759	2844751	390630	20816	3718136	
	64								
0	04 64				-1		1099	51639	
0-1	65								
<u>√</u> - 2	65				1		266	2152	
0-3	65						104	050	
Q-4	65						100	029	
#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 5-SUBSYSTEM 24 WEAPON SYSTEM

	MFG Matl	TOOL/STE MATL	SUBC	TOTAL MATERIAL	MPC	OTHER COST	SUB TOTAL
Q-1 66						8	62
TOTAL	5575		122190126	122246052	4608407	287993	135112028

#### APRIL 1972

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 5-SUBSYSTEM 24 WEAPON SYSTEM

		G	3	A	TOTAL Cost
Q-1	58				459552
Q-2	58				
Q-3	58				8955425
Q-4	58				
Q-1	59				19674492
Q-2	59				
Q-3	59				29492526
Q-4	59				
Q-1	60	411	. 81	6	22026056
Q-2	60	_			
Q-3	60	208	300	0	11124923
Q-4	60				
	61	202	16	4	11114009
	61	100			
Q-3	61	192	18	6	10534180
0 1	61	0.1	0.5	2	(050/07
	02	81	82	5	4958427
Q-2	62	0.1	~ 7	,	
	62	81	92	4	4952720
0_1	62	1 2 7	162	0	7755010
Q - 1 ∩ - 2	63	121	כנ	0	1100010
0-3	63	26	55	0	1614475
Q-4	63	20		U I	1014412
0-1	64	79	11	5	3797251
Q-2	64				
Q-3	64	1	09	9	52738
Q-4	64				
Q-1	65		-5	7	2209
Q-2	65				
0-3	65		2	3	882
Q-4	65				

II**-23**0

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 2 5-SUBSYSTEM 24 WEAPON SYSTEM

 $\left( \begin{array}{c} & & \\ & & \\ & & \\ & & \\ & & \\ & & \\ & & \end{array} \right)^{*}$ 

	GεA	TOTAL COST
Q-1 66	2	64
TOTAL	1412919	136524947

**APRIL 1972** 

ORIGINAL PAGE BLACK AND WHITE PHOTOGRAPH



### ORIGINAL PAGE BLACK AND WHITE PHOTOGRAPH



## A/V NO. 3 BOMBING/NAVIGATION SYSTEM XB-70B PROTOTYPE



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**II-23**5

SD72-SH-0003

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MAJOR AIRFRAME MATING WBS 3.0

#### GENERAL

This WBS item has been established to collect the technical, schedule and cost data relative to the joining of the major airframe. This activity presented peculiar problems on the B-70 program, primarily, Air Vehicle No. 1. A discussion of the activities and problems related to Air Vehicle No. 1 major airframe mating is contained below.

Major airframe mating commenced with the joining operation of the fuselage aft and forward center sections on May 18, 1962 and progressed to a completion of fuselage joining on July 18, 1962. (See Exhibit 1 page II-240 for pictorial display of assembly sequence.) Subsequently, major problems in structure repair, wing to fuselage mismatch and fuel tank sealing were encountered. These problems delayed completion of wing joining until April 24, 1964.

The following discussion reviews the individual problem areas which delayed the completion of mating operations on Air Vehicle 1.

#### Fuel Tank Sealing and Structural Repairs

Sealing of the fuel tanks to essentially a zero leakage level was required not only to prevent fuel leaks but primarily to prevent dissipation of the gaseous nitrogen which was used as an inerting agent in the fuel tanks. The basic fuel tank sealing was accomplished by a metal-to-metal seal provided by welding or brazing in the process of joining individual panels. The criterion for sealing required that a tank hold air at a pressure of 10 pounds per square inch with "zero leakage." Since the accuracy of the measurement of leakage was not absolutely precise, an indicated leakage of less than 10 cubic inches per minute was considered acceptable. This is equivalent to allowing one hole the diameter of a human hair in a 3,000 square foot surface.

To locate leaks, the tanks were either vacuum or pressure checked utilizing various methods of leak detection. Detecting leaks was a repetitive operation, with lesser leaks showing up as larger leaks were eliminated. Leaks which were approximately .050 to .060 inches in diameter were located by application of a soap solution over the suspect area with incoming air creating bubbles at the point of leakage. Another method utilized for detecting sizable leaks was to locate the point by an ultrasonic device which detected the passage of air through an orifice. A staining dye activated by ammonia drawn through the holes was used to outline smaller or pinpoint holes, while almost invisible holes were located by means of a helium gas sniffer.

Prior to the end of 1962 a seal brazing technique was used on the aforementioned pinhole leaks or if a doubler type repair was required. This



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WBS 3.0

technique utilized a braze alloy that melts at a temperature of about  $1,160^{\circ}$ F, applied by a hand-held torch. This process was very critical and time consuming, since no sealing was accomplished if the temperature was too low and secondary damage to the thin homeycomb panel face sheets occurred if the temperature was too high. Damage of this nature to various fuel tank sections caused a heavy workload at the end of 1962. Approximately 3000 of these areas had to be inspected and repaired where damage was found.

Another process which contributed to delay of timely completions was the portable nickel plating utilized in structural repairs. As this process was used, it was found that if a small fissure of defect existed in the panel, the plating solution could seep through into the panel and cause core corrosion. Repairing this damage necessitated the removal and replacement of complete steel honeycomb panels which in turn further delayed tank completions and sealing.

A new method sealing small pinhole leaks was developed in the early months of 1963. This process consisted of plasma-spraying aluminum over a leaking area and then applying six coats of Viton cured at 375°F. This sealing method did result in some secondary heat damage to structure but considerably less than that experienced with earlier processes. Viton sealing was constrained at times during the mating operations due to the flammability potential when welding in adjacent areas.

#### Wing Mate

Originally, a series of tools were programmed to assure the close tolerances required to mate the wing to the wing stubs. However, to conserve funds, most of the wing joining tools were eliminated in the program redirection of April 1961 and only contour support tools were provided. This necessitated matching the wing joint by optical methods, involving the fit of four contoured surfaces along an 80 foot distance to an accuracy of .008 inches or less. When this was initially attempted in late 1962, it was found that the support surfaces along the length of the wing would move and that ambient temperature differentials would also move the two surfaces of the wing and wing stub apart. After several attempts, matching by this technique was abandoned and a fitting was designed to attach to the wing stub to compensate for the inaccuracies in the joint alignment permitted by the simplified tooling. These match operations were constrained by the fuel tank sealing required in the wing root stub tanks. Fusion and electron beam welding were employed in wing joining; fusion on the inner weld and electron beam on the outer. Subsequent to inner weld completion and prior to starting the outer weld, the wing joint was x-ray inspected and the tanks were pressurized for leak detection. Extensive preparations were required in support of the external electron beam welding. These preparations consisted of cleaning the weld joint, applying RTV 77 sealant, and then placing a smooth plate over the sealant to hold the vacuum box and electron beam gun. During the early stages of wing to fuselage welding, considerable



time was spent in repairing weld areas, but this decreased as techniques improved and personnel became more experienced.

In summary, the major factors that caused delay in completing the airframe mating on the first air vehicle were as follows:

- 1. The necessity of changing the sealing process from use of a polymide sealant to application of plasma spray and Viton, a rubber based sealing compound. Fuel tank sealing, consequently installations and wing joining, were extensively delayed by this change of sealing process which required further developmental testing before air vehicle application.
- 2. Reinspection and subsequent repair of damaged areas found under the seals in brazed sections.
- 3. Replacement of sections of honeycomb panels damaged by nickel plating contamination and by gross voids that increased in size prior to completion of repairs.
- 4. Repeated pressurizing cycles necessary to detect minute air leaks.
- 5. Necessary changes to the auxiliary drive subsystem bay, involving the manufacture of an additional 6,000 individual parts.
- 6. Restrictions upon parallel effort in the air vehicle due to the inflammability of uncured Viton.
  - NOTE: For further technical data dealing with fuel tank sealing and major assembly joining see pages III-35 through III-37, WBS 1.1.

#### COST DEFINITIONS

Total costs of \$8,488,042 represent all identifiable expenditures connected with mating the major sections of the structures for Air Vehicle No. 1, 2 and a portion of 3. Installation of subsystem equipment and checkout activities are excluded. Flight test technicians and manufacturing personnel were involved in this activity and are segregated by Element of Cost (EOC). The Design/Engineering EOC contains the flight test technicians and the Production EOC reflects the manufacturing personnel.



SD72-SH-0003

EXHIBIT

## ORIGINAL PAGE BLACK AND WHITE PHOTOGRAPH



#### APRIL 1972

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 3 MAJOR AIRFRAME MATING

.

		5-SUBSYS	
		0	TOTAL
		HOURS	HOURS
		DOLLARS	DOLLARS
DESIGN/ENGINEERING		240192	240192
LABOR AT \$ 5.105		1226299	1226299
ENGR BURDEN AT \$	5.379	1292101	1292101
PRODUCTION		690140	690140
LABOR AT \$ 3.225		2225908	2225908
TEST/QC		91027	91027
LABOR AT \$ 3.644		331711	331711
MFG BURDEN AT \$	4.168	3255981	3255981
SUB-TOTAL		8332000	8332000
GEN & ADMIN		156042	156042
TOTAL COST		8488042	8488042

SUBDIVISION OF WORK COST DETAIL - SEE PAGE II-243

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#### COST BREAKDOWNS **B-70 AIRCRAFT STUDY**

4-SYSTEM 3 5-SUBSYSTEM 0

				PROD HOURS	TOTAL
				DDLLARS	DOLLARS
DESIGN/ENGIN	IEE	RING		240192	240192
LABOR AT	\$	5.105		1226299	1226299
ENGR BURDEN		AT \$	5.379	1292101	1292101
PRODUCT ION				690140	690140
LABOR AT	\$	3.225		2225908	2225908
TEST/QC				91027	91027
LABOR AT	\$	3.644		331711	331711
MFG BURDEN		AT \$	4.168	3255981	3255981
SUB-TOTAL				8332000	8332000
GEN & ADMIN		·		156042	156042
TOTAL COST	•			8488042	8488042

TIME-PHASED COST DETAIL - SEE PAGE II-244

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#### **APRIL 1972**

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING

4-SYSTEM35-SUBSYSTEM0SUBDOFWORKPRODUCTION

#### ON-SITE LABOR

	MAN- MON TH S	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q = 1 62	166.5	28359	5.143	145837	138622	284459
Q-3 62	426.0	71687	4.998	358285	362174	720459
Q = 1  63 Q = 2  63	337.5	57643	5.196	299501	310912	610413
Q-3 63	312.0	52345	5.157	269969	297643	567612
Q-1 64	177.0	30158	5.064	152707	182750	335457
TOTAL	1419.0	240192		1226299	1292101	2518400

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	PRODUCTION			
4-SYSTEM	3			
5-SUB SYSTEM	0	MAJOR	AIRFRAME	MATING
SUBD OF WORK	PRODUCTION			

#### ON-SITE LABOR

		MAN- MONTHS	LABOR HOUR S	LABOR Rate	L ABOR DOLLAR S	BUR DEN Doll Ars	LABOR + BURDEN \$
Q = 1	62	183.0	31318	3.111	97418	126368	223786
Q-3 0-4	62 62	496.5	83515	3.150	263089	353989	617078
Q - 1 Q - 2	63 63	856.5	146150	3.199	467488	630517	1098005
Q-3 0-4	63 63	870.0	146150	2.980	435583	468191	903774
Q-1 Q-2	64 64	489.0	83515	3.622	302491	423044	725535
Q-3 Q-4	64 64	742.5	130659	3.234	422486	561962	984448
Q - 1 Q - 2	65 65	246.0	42574	3.425	145822	193358	339180
Q-3	65	156.0	26259	3.486	91531	116038	207569
тот	AL	4039.5	690140		2225908	2873467	5099375

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC

4-SYSTEM 3 5-SUBSYSTEM 0 SUBD OF WORK PRODUCTION

MAJOR AIRFRAME MATING

#### **ON-SITE LABOR**

		MAN- MONTHS	LABOR HOURS	LABOR	LABOR DOLLARS	BURDEN DOLLARS	LABOR + Burden \$
Q-1	62	24.0	4130	3.292	13598		13598
Q-2	62						
Q-3	62	66.0	11016	3.331	36691		36691
Q-4	62						20071
Q-1	63	112.5	19278	3.468	66854		66854
Q-2	63			~~ ~~ ~			+C000
Q-3	63	115.0	19277	4.009	77282	211683	288965
0 - 4	63					622000	200903
0-1	64	64.5	11015	3.678	40511	52 22 8	02940
Q-2	64				10211	0000	930 <del>4</del> 9
Q-3	64	97.5	17234	3.610	62216	77076	120202
Q-4	64			5.010	022.10	11010	139292
0-1	65	33-0	5616	3.702	21300	25710	44710
0-2	65		0010	2010	21000	27410	40110
$\bar{0} - 3$	65	21.0	3461	3, 931	12250	14000	20250
			0401	J. 0 J.	13233	14333	28228
тот	AL	533.5	91027		331711	382514	714225

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM	3	NA TOD A TOPDAME MARTING
5-SUBSYSTEM	0	MAJOR AIRFRAME MAIING
SUBD OF WORK	PRODUCTION	

		MAN- Mon Ths	LABOR HOUR S	LABOR Rate	LABOR DOLLARS	BUR DEN Dollars	LABOR + BURDEN \$	G&A
Q-1 Q-2	62 62	373.5	63807	4.025	256853	26 <b>49</b> 90	521843	8789
Q-3 Q-4	62 62	988.5	166218	3.959	658065	716163	1374228	23066
Q-1 Q-2	63 63	1306.5	223071	3.738	833843	941429	1775272	29683
Q-3 Q-4	63 63	1297.0	217772	3.595	782834	977517	1760351	29433
Q-1 Q-2	64 64	730.5	124688	3.976	495709	659132	1154841	24572
Q-3 Q-4	64 64	840.0	147893	3.277	484702	639038	1123740	23911
Q-1 Q-2	65 65	279.0	48190	3.468	167122	218776	385898	10296
Q-3	65	177.0	29720	3.526	104790	131037	235827	629 <b>2</b>
TOT	AL	5992.0	1021359		3783918	4548082	8332000	156042

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

MAJOR AIRFRAME MATING

4-SYSTEM 3 5-SUBSYSTEM 0 SUBD OF WORK PRODUCTION

	TOTAL
Q-1 62	530632
Q-2 62	
Q-3 62	1397294
Q-4 62	
Q-1 63	1804955
0-2 63	
Q-3 63	1789784
Q-4 63	
Q-1 64	1179413
Q-2 64	
Q-3 64	1147651
Q-4 64	
Q-1 65	396194
Q-2 65	
Q-3 65	242119
TOTAL	8488042

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#### WBS 4.0: FLIGHT TEST

#### GENERAL:

The flight test program was conducted on two XB-70 air vehicles and was a flight research program which demonstrated the design, fabrication, and technical feasibility of long range Mach 3 flights at high altitude. It was the final phase in a very large step forward in the state-of-the-art of manned aircraft design achieved during the B-70 development program. The advanced aerodynamic and high temperature design concepts incorporated in the XB-70 air vehicles were all essentially verified during the flight test program. These state-of-the-art achievements as well as the XB-70 configuration, are defined in the air vehicle subsystems technical descriptions presented in Volumes III and IV of this report.

The flight test program was conducted to accomplish the established detail objectives as presented by Exhibit 1, page II-253. The air vehicle configuration, allowable configuration changes, and all flight test support requirements were based on that deemed necessary to safely accomplish the test objectives. All effort, including that involved with manufacturing checkout, preflight, and the flight test program, was scheduled to achieve the desired goals by the most timely and economical means. In this same framework, the flight test program management was streamlined to provide closer control and to reduce reaction time. Exhibit 2, page II-254, presents the management matrix employed, and as shown, operations control was based on the Joint Test Force (JTF) concept while contractual coverage was provided by the Development Research Vehicle (DRV) Division of Wright Patterson Air Force Base.

This section presents a technical description of the effort expended during final operations, preflight, and the flight test program. The support items during this program phase, such as, Flight Test Ground Support Equipment (FTGSE), Special Test Equipment, and Tooling are discussed under WBS Sections 5.0, 7.0, and 8.0, respectively. The comparison between the flight test program planned for the RS-70 and that conducted on the XB-70 air vehicles is presented under Air Vehicle Summary Data; WBS: 1.0. The Air Vehicle Summary Data section also presents the XB-70 operational flight limits and flight testing hours expended per air vehicle major subsystem.

#### COST DEFINITION

The recorded costs displayed in this WBS item are accumulated at levels 5 and 6 in the following manner:



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WBS 4.40	Final Operations and Preflight	\$ -0- Page II-259
WBS 4.41	Flight Test	\$ 2,781,972 Page II-310
WBS 4.41.1	Flight Operations	\$29,594,412 Page <b>II-310</b>
WBS 4.41.2	Air Vehicle Maintenance	\$ 5,216,014 Page <b>II-310</b>
WBS 4.41.3	Instrumentation	\$ 1,164,316 Page II-311
WBS 4.41.4	Post Flight Evaluation	<b>\$ -0- Page II-</b> 311
WBS 4.41.5	Major Air Vehicle Repair	<u>\$ 496,336</u> Page II-311
	Total WBS 4.0 Flight Test	\$39,253,050

Definition and explanation of the cost data can be found on the pages indicated above.

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## **APPROVED PROGRAM OBJECTIVES**

- DEMONSTRATE TECHNICAL FEASIBILITY OF ACHIEVING THE PERFORMANCE GOALS OF A M3.0 AIR VEHICLE
- DEMONSTRATE FLYING QUALITIES OF THE XB-70 UNDER FAIR WEATHER CONDITIONS
- DEFINE MAJOR PROBLEMS
- ESTABLISH AIRWORTHINESS & ACQUIRE DATA ON 15 SUBSYSTEMS
- QUANTITATIVELY EVALUATE WITH DATA:

STABILITY & CONTROL PERFORMANCE VIBRATION / FLUTTER / ACOUSTICS STRUCTURAL INTEGRITY TEMPERATURE EFFECTS

- ACHIEVE M1 THE FIRST FLIGHT
- ACHIEVE 30 MIN SUSTAINED M3 DURING PROGRAM
- OBTAIN 180 HOURS OF FLIGHT IN 22-1/2 AOM'S
- PRESENT ANALYSIS TO GOVT IN MONTHLY PROGRESS REPORTS

II -253

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# **XB-70 FLIGHT TEST PROGRAM**



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SD72-SH-0003 EXHIBIT 2



#### WBS 4.0 FLIGHT TEST

#### FINAL OPERATIONS AND PREFLIGHT

WBS CODE: 4.40

#### GENERAL:

In a production type program, final operations and preflight are performed during two separate phases with final operations and checkout being the last phase of manufacturing and preflight the initial phase of Engineering Flight Test. However, during the XB-70 program, to achieve the most efficient programming of resources available and to reduce calendar time, final operations and preflight were performed concurrently on both Air Vehicle No. 1 and Air Vehicle No. 2.

The No. 1 air vehicle went into the concurrent phase of final operations and preflight on May 12, 1964. During the next three months certain manufacturing items were completed, systems operational evaluations, Phase I proof loads, and flutter and vibration tests were conducted with NR and AF inspections completed on August 7, 1964. Taxi tests No. 1, No. 2 and No. 3 were conducted on August 9th, 16th, and 24th, respectively. These taxi tests were unsatisfactory due mainly to hydraulic line failures. Taxi test No. 4 was conducted on August 25th and taxi tests No. 5 and No. 6 were conducted on September 14th with the first flight on the No. 1 air vehicle occurring on September 21, 1964. Subsequent to the fourth flight on October 12, 1964, the No. 1 air vehicle went into the second phase of final operations and preflight in preparation for high supersonic flight. During this time period, the two major items were the completion of the Air Induction System and Phase II proof loads; flight No. 5 occurred on February 16, 1965.

The final operations and preflight on air vehicle No. 2 were initated on May 29, 1965 with taxi tests conducted on July 10, 1965. The first flight on Air Vehicle No. 2 occurred on July 16, 1965.

#### DISCUSSION:

On Air Vehicle No. 1 only, final operations and preflight were performed in two phases with the first phase consisting of that testing necessary to assure safe flight up to Mach 1.4 while during the second phase the air vehicle was prepared for high supersonic flight. The systems ground checkout, evaluation, and calibrations were scheduled in this manner to be compatible with the Air Induction Control System (AICS) schedule (see WBS 1.5) and to provide an advance overall "look" at the B-70 in flight for possible basic problem areas. The final operation and preflight requirements were essentially the same for both air vehicles except for structural loading and dynamic tests which were conducted on Air Vehicle No. 1 only.

The No. 1 air vehicle was "rolled-out" from Manufacturing on May 11, 1964 and was transferred officially to Engineering Flight Test on May 12, 1964, which was the official start date for final operations and preflight. The initial period was devoted to subsystems operations which consisted primarily of final operations on the hydraulic systems, landing gear, and first phase of operations on the flight controls. Installation of J-93



engine modifications for the afterburner fuel pumps and throttle linkage were also accomplished during this period together with final fuel tank installations and pressure tests. After the initial period, engineering ground tests required during the first phase of preflight were conducted. These tests consisted essentially of subsystem evaluations and calibrations including instrumentation checkout and calibrations.

The initial engineering evaluations were for Phase I proof loads and dynamic tests on the airframe structure. The proof load tests consisted of applying simulated air loads to the main gear, nose gear, and all alighting and arresting subsystem doors during subsystem operations. The structural dynamics tests consisted of vibration tests of the air vehicle to investigate the flutter modes and to verify calculated data as required to assure safe flight operation. Exhibits 3, 4, 5, and 6. on pages II-260. II-261. II-262. and II-263, respectively, present the test setup for the vibration and flutter ground tests. Following the structural dynamics tests, engineering type tests were conducted to verify subsystem operation and subsystem interface functions to ensure safety of flight. During these preflight tests, engineering data were obtained by the airborne instrumentation system for utilization during the flight testing.

During the initial preflight phase, the air vehicle was weighed (before expendables were added) to establish the empty gross weight and center of gravity of the air vehicle, see Exhibit 7, page II-264. Landing gear tests were initiated and partially completed; however several components were short which delayed completion of tests until after taxi tests. Exhibit 8, page II-265, presents a view of the main gear in the initial phase of retraction. Fuel system calibrations were accomplished to verify quantity and operation of the gaging system, fuel supply, fuel management, cooling fuel flow loop, transfer system, and to establish the unavailable fuel of individual tanks and total fuel system. The emergency environmental system, which used ammonia and required the air vehicle to be outside the hangar with special GSE, was evaluated to assure correct operation, verify servicing techniques/equipment, and to establish zero leakage.

In addition to the engineering tests described above, specific subsystem tests conducted required various combinations of engine operations. These engineering tests were identified as propulsion and integrated systems tests which confirmed operation of the propulsion subsystem, secondary power subsystems, and the ECS. These single-engine and multi-engine ground operations also provided vibration data of the engine, structure, and secondary power subsystems. During the engine runs and with special instrumentation located at various points, both near and far field noise level data were obtained as part of the NASA Noise and Sonic Boom Program.

The No. 1 air vehicle was removed from flight status after Flight No. 4 and the second phase of final operations and preflight was performed. This phase of ground tests was in preparation for high supersonic flight with the major effort directed toward the satisfactory completion of the Phase II proof load tests and the installation, checkout, and preflight of the

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AICS. In addition to this effort, deficiencies disclosed during the initial flights, which would interfere with flight objectives, were corrected and the flight augmentation control system (FACS) was preflighted (this system was not activated during first flights).

The Phase II proof loading tests were conducted to demonstrate that major surfaces would adequately perform, both statically and dynamically, under the high loads of high speed flight; Exhibit 9, page II-266, presents the areas tested. As shown, the areas loaded were the folding wing tips, elevons, canard surfaces and flaps, vertical stabilizers, AIS ramps and bypass doors, flight control system, and the emergency rain air scoop. Exhibits 10, through 14, pages II-267 through II-271, present actual test setups typical of those employed during the proof testing. As shown as by the exhibits, the surfaces were loaded by hydraulic actuators which transmitted the load through vacuum pads attached to the surfaces: the exception to this was the loading of the AIS ramps where the inlet was pressurized. During the tests, structural stiffness and deflection data were obtained and noninterference loaded surface operation verified. The maximum proof loads applied were 80% of design limit except for specific areas, such as the control column, cables, elevons, where 100% design limit load was applied.

The Phase II proof tests of the AIS was initiated on the left-hand inlet; however, due to a failure, the proof testing of the AIS was completed on the right-hand inlet which required repositioning of the special test setup. The failure experienced was at the forward hinge of the first movable ramp at approximately 90% load. The failure also caused secondary damage to local structure behind the ramp in the nose wheel well area. The beefup of the ramps and the repair of structural damage slipped Flight No. 5 schedule by approximately one month.

The following table presents the load level applied during the Phase II tests:

Load Level

Item

Static Loading:

Control System	Limit pilot loads
Horizontal Stabilizer	80% limit
Wing Folding Tips	80% limit inboard wing 64% limit movable wing
Elevons	80% limit
AIS Duct and Ramps	100% limit
AIS Bypass Doors	100% limit

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Space Division

North American Rockwell

Item

Operational Tests Under Load:

Load Level

Horizontal Stabilizer	72% limit
Vertical Stabilizer	80% limit
Wing Folding Tips	100% limit operational load
Elevons	80% limit
AIS Ramps (one side)	100% limit operational load
AIS Bypass Doors	100% limit
Emergency Rain Scoop	100% limit

Taxi Tests: The first taxi test occurred on August 9, 1964 where problems with nose wheel steering, brake chatter, loss of Primary No. 1 hydraulic system, and ammonia fumes in the cockpit caused the tests to be aborted. The nose wheel steering problem was two fold: one was an apparent drift and the other was an apparent lack of authority. The control valve of the steering unit was replaced which fixed the drift, in addition, an indicator was installed in the cockpit to provide the pilot nose wheel position to aid primarily in engagement of steering. The authority of the nose wheel steering was squawked throughout the flight test program; however, the condition was minimized to some extent by pilot technique.

The brake chatter was believed to be caused by the brake linings not being "burned-in" (seating the discs). However, a switch was installed so that fore and aft brakes of each bogie could be selected so that "burn-in" could be accomplished during the next taxi. The Primary No. 1 hydraulics was lost due to a line break at a coupling in ADS Bay No. 1. (for a summary of hydraulic line failure technical analyses, see Technical Driver: Hydraulic Pumps, WBS 1.4.) The ammonia fumes in the cockpit were caused by contaminated make-up air from the engines which had ingested the overboard exhausted ammonia fumes. The condition could only happen during taxi and ground operations where there was insufficient wind to clear the area forward of the inlets of fumes. To correct the unacceptable condition, a ground lockout relay was installed which closed the make-up air valve during ground operations.

The second taxi test was made on August 16th with the first pass to 35 KIAS satisfactorily completed. However, the second pass was aborted due to a hydraulic leak caused by a ruptured line at a manifold fitting in ADS Bay No. 1. During the taxi tests at low speed, the brakes still chattered which caused the landing gear doors to open and the No. 3 engine nozzle to go full open. The "burning-in" of the brake linings was scheduled to continue during the next taxi tests.

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The third taxi tests were conducted on August 24th with two passes up to 65 KIAS satisfactorily completed. However, after the second pass a hydraulic leak was found in ADS Bay No. 6 which was caused by a primary pump valve manifold seal failure. It was subsequently found that the lefthand brake actuator was also leaking which resulted in a rework of installing a new viton seal in all brake actuators. On August the 25th the fourth taxi tests were satisfactorily completed with no problems encountered. Two passes were made; the first up to 86 KIAS (75 knots ground speed) and the second up to 122 KIAS (100 K ground speed). During the second pass, the drag chute was satisfactorily deployed at 122 KIAS and jettisoned at 80 KIAS with subsequent deceleration accomplished by moderate to heavy braking.

The fifth and sixth taxi tests were satisfactorily completed on September 14, 1964. The fifth taxi test consisted of one pass at 87 KIAS while the sixth consisted of one pass at 97 KIAS. After the sixth taxi test, all eight main gear tires were replaced (due to wear) and the landing gear cycled 20 times in preparation for first flight. In addition, several binding hydraulic valves were replaced and the No. 4 and No. 6 engines had to be replaced due to foreign object damage (FOD).

The extensive prior-to-first flight taxi tests were conducted to provide the pilot (and engineering data) the capability of evaluating the low and high speed characteristics of the new air vehicle. This included radius of turn, steering, braking, intermediate speed directional control, high speed directional control and nose wheel "lift-off" or longitudinal control power. It also allowed the pilot to evaluate visibility and "get-the-feel" of the air vehicle, such as establishing reference points, during these critical phases of flight. Since these conditions were evaluated on Air Vehicle No. 1, the No. 2 air vehicle taxi tests were completed in one day; July 10, 1965. For taxi schedules and taxi area plots, see Exhibits 14A and 14B. on pages II-272 and II-273.

#### COST DEFINITION:

Cost of the final operations and preflight activities performed on A/V's 1 and 2 are contained in WBS 1.12 (Volume IV, page IV-647). Flight test personnel performing final operations and preflight activities cannot be segregated from the manufacturing personnel performing systems installation and checkout on the two air vehicles. This condition exists because of the uniqueness of the program. If the B-70 had been a production program instead of a two-vehicle research and development program, manufacturing and flight test activities would have been two separate, independent operations. However, because of budgetary and schedule constraints, maximum utilization of personnel was essential. Therefore subsystem installation, checkout, final operations and preflight activities were performed by both manufacturing and flight test personnel concurrently. No segregation of activities in this area is available because the same accounting procedures were utilized by all personnel.

SD72-SH-0003

## STRUCTURAL DYNAMICS GROUND VIBRATION TESTS

3

- TEST EQUIPMENT AVAILABLE AND ON SITE
- TEST SETUP INITIATED

EXCITATION SYSTEM
DATA ACQUISITION SYSTEM

(3) AIR SUSPENSION SYSTEM

INSTRUMENTATION VAN

2

(1)

A/C

(c)

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**PHASE II PROOF TESTS** PROOF AND OPERATIONAL TESTS UNDER LOAD





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# FINAL OPERATIONS AND PREFLIGHT TAXI TESTS, AIR VEHICLE 1

WBS 4.40 MAX, SPEED TEST KNOTS (IAS) MILES 1/ 1. TEST ABORTED DUE TO BREAK IN ADS BAY 1 20 1.7 HYDRAULIC LINE, STEERING PROBLEMS, 35 2.3 BRAKE CHATTER, AND NH3 FLUMES IN COCKPIT 40 4.7 65 4/ 86 FIRST RUN: EXPERIENCED BRAKE CHATTER, 9.4 122 LANDING GEAR DOOR MALFUNCTION, AND 5 97 5.2 UNPLANNED OPENING OF NO. 3 ENGINE NOZZLE. SECOND RUN: CANCELLED DUE TO HYDRAULIC LEAK. 87 3.2 POST-RUNS INSPECTION REVEALED HYDRAULIC LEAK IN ADS BAY 6 AND LEFT-HAND BRAKE ACTUATOR SEALS REQUIRED REPLACEMENT. 4 FIRST RUN: NOSE WHEEL STEERING EVALUATION COMPLETED. SECOND RUN: DRAG CHUTE JETTISONED SATISFACTORILY AT 80 KIAS AFTER BEING DEPLOYED AT 122 KIAS. BRAKES OPERATED SATISFACTORILY AND NO HYDRAULIC LEAKAGES WERE ENCOUNTERED. SUCCESSFUL TESTS 10 11 12 13 14 15 16 17 18 19 20 21 22 23 24 25 26 27 28 29 30 31 1 2 3 9 10 11 12 13 14 15 16 17 9 8 4 5 6 7 AUG 1964 SEP

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EXHIBIT

14A





# PREFLIGHT & FINAL OPERATIONS (AV NO. 1)

WBS 4.40

Prior to conducting the air vehicle rollout on 5-11-64, it had been resolved that the following manufacturing operations would be deferred for later completions:

Hydraulic servicing and power-on

Primary and utility cross systems leak check

Landing gear operations

By-pass door system

Flight Control, AICS, brake control, steering, wing fold, preliminary and complete fuel system functional operations

Fire Protection

Thrust control and speed reset

Engine ignition ground and air start

Engine and ADS oil indicating

Engine windmill brake

Emergency generator

Drag chute

Windshield ramp

Flood flow scoop

Auxiliary cooling

Escape system

Secondary pressure and cabin pressure leakage test at 5 psi

WBS 4.40 ROLLOUT FINAL OPERATIONS AND PREFLIGHT EMERGENCY GENERATOR **AIR VEHICLE 1** PHASE I PROOF LOADS TESTS FIRE PROTECTION SYS OPS DRAG CHUTE OPS WINDSHIELD RAMP OPS FLOOD FLOW SCOOP OPS WING FOLD OPS ESCAPE SYS OPS FUEL TANK PRESSURIZATION TRIM CONTROL AND FACS, FUNCT AND OPS FUEL TANK WORK FLT CONTROL SYS FUNCT AND OPS AICS - BYPASS DOORS GROUND VIBRATION TEST SETUP ADJUST LOAD CELLS/AIR VEHICLE - CHECKOUT SHAKERS REMOVE NOSE WHEELS/BOGIES - POS LOAD CELLS GROUND VIBRATION TESTS II-275 SEC PRESSURE (SEALS) AND CABIN PRESSURE (5 PSI) INSTALL NOSE WHEELS/BOGIES COMPLETE FUEL SYS FUNCTION NOSE WHEEL STEERING OPS FLUID POWER LEAK CHECKS WEIGH AIR VEHICLE FUEL SYS CALIBRATION AND FLUSHING SYS INSTL AMMONIA FLUSHING BRAKE SYS OPS ELECTRICAL-ELECTRONIC COMPATIBILITY SD72-SH-0003 SINGLE ENGINE RUNS AND FOD SWEEP PROPULSION AND INTEGRATED SYS TESTS PRE- AND POST-TEST OPS ELECTRICAL SYS PHASING FLUID POWER SURGE MULTI-ENGINE RUNS INSTRUMENTATION INSTL AND CHECKOUT POST-AND PREFLIGHT AND SERVICES TAXI TESTS REWORK LANDING GEAR AND GEAR OPS NAA AND AF FINAL INSPECTION INSTRUMENTATION AND CALIB POST- AND PREFLIGHT AND SERVICES FIRST FLIGHT 77 15 12 29 5 19 26 3 10 17 . 24 31 7 14 21 28 4 11 . 18 25 2 JUN JUL AUG SEP

## Final Operations and Preflight Air Vehicle 2

WBS 4.40





### FLIGHT TESTING

WBS CODE: 4.41

The flight testing development and research phase of the B-70 Flight Test Program was initiated on September 21, 1964 with the first flight on Air Vehicle No. 1. The flight test development program was concluded with Flight No. 49 on Air Vehicle No. 1 on May 9, 1966 and Flight No. 46 on Air Vehicle No. 2 on June 8, 1966. The No. 2 air vehicle was lost on Flight No. 46 as a result of a midair collision between an F-104 chase plane and the B-70. The NASA/Air Force joint contracted research phase continued on Air Vehicle No. 1 as described under Air Vehicle Summary Data; WBS 1.0 in Volume II of this report. On Flight No. 83 on February 4, 1969, the No. 1 air vehicle was ferried to Wright Patterson Air Force Base and turned over to the Air Museum.

The flight test development program performance was based on Aircraft Operational Months (AOM) which is discussed under Air Vehicle Maintenance; WBS 4.41.2. The development program consisted of 22-1/2 AOM's or 35 total aircraft calendar months over a time span of approximately 23 calendar months. Exhibits 15 and 16, on pages II-285 and II-286, present the flight rate and flight hour performance showing the comparison between scheduled and actual. As depicted, the total number of flights was 95 and the total hours was 186:23 (scheduled 180 hours). Exhibits 17 and 18, pages II-287 and II-288, present the turnaround performance showing the program goal versus actuals for unscheduled and scheduled maintenance based on AOM's. (Scheduled and unscheduled maintenance are also discussed under WBS 4.41.2).

Exhibit 19, page II-289, presents the XB-70 flight envelope showing, by solid line, that area explored by Air Vehicle No. 1 and, by a dash line, that explored by Air Vehicle No. 2. During these explorations, data were obtained on the Air Vehicle and its subsystems as depicted by Exhibit 12, page II-22, under Air Vehicle Summary Data; WBS: 1.0. A general summary of the data parameters is presented in Exhibit 20, page II-290. As shown, during the test program there were 249 parameters deleted and 378 parameters added resulting in a growth from 837 at onset to a total of 966 parameters. (It should be noted that the exhibit presents a general summation of a most variant condition that existed on both air vehicles).

In subsequent subsections, as identified by the WBS, narrative discussions are presented on Flight Operations, Air Vehicle Maintenance, Instrumentation, Post Flight Evaluation, and Major Repairs. In addition, the following paragraphs present the Pilot's Summary which was extracted in total from "XB-70A Pilot's Summary Report" written by Mr. A. S. White, Chief Test Pilot, Engineering Flight Test, North American Rockwell. In conjunction with the pilot's summary, the following exhibits are presented: Exhibit 21, page II-291, XB-70 at lift off; Exhibit 22, page II-292, March 3 radar ground plot; and Exhibit 23, page II-293, XB-70 during the landing phase.



### Pilot's Summary Report:

#### Summary:

The XB-70 has proven itself to be a very remarkable airplane. It has accomplished every milestone that was set down for it. No aerodynamic changes were required in order to achieve the objectives; but in spite of its tremendous performance, the XB-70 is an unfinished airplane. It was operated in a completely new speed and altitude range, but with off-theshelf navigation equipment and flight instrumentation that were obsolete for this type of flight operation. This pilot believes that with normal development, including some aerodynamic refinement, some system changes, and better instruments and navigational equipment, this would be a truly outstanding airplane.

#### General:

In reading this pilot's summary report, consideration should be given to the fact that the combination of this airplane's size, weight, and speed ranges compares to no other airplane in existence; and that the program was primarily a research program with ground rules that allowed only those changes which were necessary to safely accomplish the test objectives. In the normal sequence of development to an operational airplane, many of the opinions and recommendations expressed herein would have been made after the initial airworthiness flights. They are presented now, for the record, and for consideration if improvements can be made for follow-on programs.

In general terms, the XB-70A was an interesting airplane to fly. It had some peculiarities due to size, weight, and configuration that were different from most other airplanes. The movable wing tips introduced some new characteristics in the airplane that had not been experienced before, such as the wide variation in directional stability, roll power, and dihedral effect.

The airplane had a tremendous performance capability and was a thrill to fly from that standpoint, but this capability combined with the fact that the airplane was climbed and cruised very near the boundary of the allowable flight envelope quicker and easier than any other airplane this pilot has flown. The duct pressure limits and the low allowable "g" aggravated the recovery. All this adds to the pressure on the pilot by requiring greater concentration on his part. This was acceptable in a research program of this nature, but would certainly not be acceptable in an operational vehicle. The potential for a great airplane is here, but some refinements are required. More specifically, the pilot's opinions and recommendations are as follows:

### Ground Handling Qualities:

Taxiing the airplane was not a difficult operation. Turns from one narrow taxiway to another narrow taxiway was made with acceptable accuracy. However, parking the airplane in a precise location was difficult because of



the restricted visibility, configuration of the airplane, and lack of repeatability of the nosewheel steering system, i.e., the nosewheel steering rate varied with the load on the nose gear so that full rudder pedal application did not consistently give the same turning rate. Ground observers were required for parking, but their inputs were sometimes valueless due to the steering system's inability to follow their directions. Cooper Rating of 3.5. See Exhibit 24, page II-294, for clarification.

Originally the braking system was unsatisfactory at very low speeds, but with development, the system became marginally satisfactory. The airplane could be stopped without brake chatter as long as the pilot anticipated far enough in advance to be able to apply very light braking to make the stop. If an abrupt stop was required at low speeds, heavy brake chatter occurred. Cooper Rating 4.5.

### In-Flight Handling Qualities:

Low Speed: The pitch control in low speed flight regime was very good. Some lack in airplane response could be detected during abrupt pitch maneuvers; however, the response was satisfactory in all of the normal maneuvers used in flying the airplane. The long period of the short-period oscillation was different from most other airplanes, but did not cause difficulty in controlling the airplane. The damping in pitch was good in the subsonic flight regime. Cooper Rating of 2.5.

The main difficulty in flying the XB-70 was caused by a combination of characteristics in the lateral-directional sense. These characteristics were: First, that the period of the short-period oscillation was very long as compared to most other airplanes, secondly, the side force per degree of sideslip was low in this airplane. This combination made it difficult for the pilot to sense inadvertent sideslips. In addition to those characteristics, the airplane had a marked positive dihedral effect (particularly #2 A/V) and it had excessive adverse yaw due to ailerons. Considering these four characteristics together, if an inadvertent sideslip occurred without the pilot's knowledge (busy with other system operation), the airplane rolled due to the dihedral effect. Instinctive reaction of the pilot was to counteract the roll with aileron; however, the aileron input increased the sideslip because of the high adverse yaw due to aileron. This increased sideslip caused more roll away from the aileron input and aggravated the situation. The solution was that the pilot had to fly the sideslip or yaw indicator religiously throughout the flight regime. This general characteristic of the XB-70 was most predominant with the wing tips up. Cooper Rating of 4.5.

Placing the wingtips in the one-half position reduced the tendency for the situation explained in the previous paragraph, because lowering the wingtips reduced the roll power by approximately 50%, therefore reducing yaw due to aileron. In addition, the tips down configuration reduced the positive dihedral effect. Cooper Rating of 4.0.



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Visibility with the windshield in the up position was unsatisfactory. The pilot cannot see the horizon ahead and must make the flight almost entirely with reference to the pitch attitude indicating system. The poor quality of the heading information, precise heading, climb schedule made altitude flying extremely difficult. In addition, images of light-colored ground such as snow and sandy desert areas reflected badly in the windshield during turns, at times causing vertigo. Cooper Rating of 5.0.

<u>Unstarts</u>: The inadvertent inlet unstarts encountered in the flight test program varied in intensity from mild to severe. They were breath-taking to say the least. In the case of a severe unstart, it jarred the airplane rather violently and was followed by heavy buffeting, intense aerodynamic noise, and minor trim changes. At Mach 3, the primary trim change was in roll, but usually did not persist since the other inlet would normally unstart within a few seconds. If the inlet system did not effect an immediate restart, the inlet would go into buzz. The buzz cycle was immediately recognizable to the pilot since it was almost purely a lateral oscillation at about the natural frequency of the fuselage. If it was not corrected, it built up in intensity to a very disturbing, if not destructive, magnitude. In spite of the severity of the transients caused by inlet unstarts, airplane control was considered good.

Landing: The XB-70 was not a difficult airplane to land. Some care had to be used due to the distance between the pilot and the main gear and due to the crew station height above the main gear at touchdown. Because of these dimensions, it was easy to undershoot the runway. After some practice, all pilots were able to make satisfactory landings without external assistance from chase aircraft.

The wing of the XB-70 experienced a strong ground effect in the proximity of the runway which helped considerably in making relatively smooth landings. The secret to a good landing, like in most airplanes, was a good stabilized approach using a rather low rate of descent (2 to 3 degrees glide scope).

Crosswind landings in the XB-70 were not as difficult as was predicted. The relatively shallow bank angle per degree of sideslip made the wing-down technique rather easy to use with the XB-70.

Landing in turbulent air required additional pilot concentration, primarily due to previously mentioned relationship between inadvertent sideslip, dihedral effect, and yaw due to ailerons. The pilot was required to watch the yaw indicator very closely in making an approach so as not to get into this inadvertent sideslip condition close to the ground. Landing Cooper Rating was 3.5.

Cockpit: Controls and Layout:

Capsule: The pressure seals in the capsule doors were torn loose many

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times when the pilots entered the capsules. Entering the capsule, particularly when wearing the pressure suit, was very difficult due to the lack of space. The seals should be guarded so that they are not damaged in this way.

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The original installation of the capsule handgrip seat pins included a lanyard and a take up reel which were installed behind the pilots shoulder. A large percentage of the time the take-up reel did not operate; and when it did operate, the pin was extremely hard to reach to reinstall after flight. A temporary fix was made by cutting the lanyard, thereby eliminating the take-up reel. This worked **s**atisfactorily, except for the minor inconvenience of not having a place to stow the pin.

The emergency parachute and riser cutter handles and the hinge split handle were difficult to see when encapsuled. When the pilot raised his head to see the handles with his helmet on, he had to raise his helmet with his hand in order to see past the bow of the helmet. This was particularly true when the pilot was wearing a pressure suit and was aggravated under the dynamic conditions encountered after ejection.

The pressurization and capsule oxygen gages were particularly hard to see once in the capsule. The safety belt was almost impossible to adjust after it has been fastened. The seal deflate button was very difficult to actuate due to its location when the pilot was wearing a pressure suit.

The hot mike interphone capability during encapsulation was operable only after the capsule doors were closed. It appears wise, after the experience of the recent ejection, that the hot mike be actuated in another way in addition to the door closure. It should be connected to the handgrips so that when encapsulation is made, the hot mike interphone is available even if the doors are not closed.

The manual impact attenuator inflation device was extremely difficult to get to while encapsulated. It was recommended that some thought be given to relocating this device between the pilot's knees for easier access.

The Secondary Nozzle Rheostat: This rheostat is in a poor location considering the number of times it was used during flight. It would be desirable to move it forward in the area near the oxygen and visor heater switches.

Hydraulic Pump Status Indicators: The hydraulic pump status indicators became almost useless in view of the number of times the pump status indicators showed yellow with the pumps operating properly.

Nosewheel Steering System: The nosewheel steering engage button ideally should be mounted on the control wheel. Originally this was not done because of lack of space on the wheel. The pilots believed that the augmentation disengage switch should have the priority location on the wheel. The experience gained in the flight test program indicates that it would be satisfactory to move the augmentation disengage switch to the same area as



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the augmentation engage button on the console and put the nosewheel steering engage switch on the control wheel.

TACAN: The TACAN instrumentation on the XB-70 was only marginally satisfactory. Early in the program the pilots were requested to list the minimum equipment with which they could accomplish the mission, and one TACAN was suggested for navigation. Experience has now shown that without radar tracking and the occasional assistance of the FAA centers, some of the missions would have been extremely difficult to complete satisfactorily due to the quality of this equipment and installation.

Attitude and Heading Information: The attitude and heading information was marginally satisfactory. This added to the pilot's difficulty in navigating the airplane. In view of the above two conditions, it was highly desirable to have a more reliable navigation system. An inertial platform was highly recommended.

AICS Controls: The AICS controls were satisfactory for the development stages, although the controls in A/V #1 were minimum satisfactory. It was highly recommended that this system be reviewed for future use.

Flaps: The flap system was marginally satisfactory. Because of a design problem, the flaps were to be raised for the taxi and lowered just prior to takeoff in order to ensure that they would retract in flight. Occasionally they would not extend for landing, and occasionally they would not retract after landing. The result was that the pilots lost confidence in the flap system. It was recommended that a design change be made to eliminate the necessity of the intricate procedure that was required to operate the flap system.

<u>Map Case</u>: The map case was almost inaccessible to the pilots. In most cases the pilot had to get out of his seat to get the equipment from the map case. Although the map case was of little use to the pilots, it was the only place in the cockpit that extra material could be stored. Check lists and pilot's data cards had to be strapped to the pilot's legs to be of any use. Some consideration should be given to a more convenient stowage space for let-down charts, handbook, and additional maps, particularly if this airplane is to be used for cross-country work in the future.

Wingtip Selector Switch: If the flaps are lowered for landing prior to raising the wingtips to the up position, it would be possible to lose control of the airplane. It was recommended that a safety device be installed to prevent lowering the flaps when the tips were not in the up position.



The airplane had moderate buffet at low speeds with the gear and flaps down. There was a minor change in the buffet level as the gear and flaps were raised. Some buffet persisted and a high aerodynamic noise level existed until the airplane accelerated to .87 Mn or above, at which time the buffet completely disappeared and the noise was reduced to at least half of the low-speed level.

High Speed: Pitch control during the transonic acceleration and low supersonic speed ranges was very good. Above Mach 2 it became evident that there was some deterioration in the effectiveness of the pitch control. At speeds above 2.5 Mn, the force level required to maneuver the airplane was excessive due to this deterioration and due to the action of the "g" bellows. Cooper Rating of 3.5.

The directional control was adequate to take care of three engines out on one side, but was much less effective in producing yaw than were the ailerons. Considering the capability of the rudders alone, they were considered adequate. Cooper Rate of 3.0. However, the powerful capability of the ailerons in producing yaw was considered the most objectionable characteristic in the airplane. Cooper Rating of 4.5.

The roll response of the airplane in the supersonic flight regime was good. The roll response did not seem to deteriorate at the same rate as pitch response, and therefore there was less compatibility in the response of the airplane between Mach 2 and 2.7 than in other areas. As the airplane approached Mach 3, the roll response deteriorated and the yaw due to aileron input was reduced; therefore the handling qualities of the airplane in roll were better at Mach 3 than at any other supersonic speed. Cooper Rating for the roll control system was 3.5.

Although there were minor differences in the flight characteristics between 1.4 and Mach 3 with all augmentation off, the general characteristics were the same. The short-period oscillations in pitch and yaw were four to six seconds in length and the damping in pitch and yaw was relatively poor. The airplane could be flown at all speeds with the augmentation off, except that extreme care had to used in the use of ailerons because of the strong tendency to excite lateral-directional oscillations with the ailerons. This was due to a high roll rate in the unaugmented case and because of the excessive level of yaw due to ailerons. When the pilots left the ailerons alone, the yawing oscillations would damp. Unaugmented supersonic flight: Cooper Rating of 5.0.

<u>Visibility</u>: The visibility with the windshield down was satisfactory. Although some forward visibility was lost after the rotation at takeoff and during the initial part of the climbout, the remainder of the subsonic flight was satisfactory.

The visibility for landing was considered good. At no time was the pilot aware of any loss of the runway visibility during approach, flare, and touchdown: Cooper Rating of 3.0.



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The longitudinal trim system was very good. Cooper Rating of 2.0.

The lateral trim system was also very good, except that the primary lateral trim control was difficult to operate with a high degree of accuracy when the pilot was wearing heavy gloves. Cooper Rating of 3.0.

The directional trim system was very good after the gear had been retracted. It was too sensitive with the gear down, which caused the pilot to overcontrol when attempting to trim out directionally. Occasionally some difficulty was encountered when engaging the nosewheel steering system due to the inability of finding the neutral directional trim position.

For instance, the pilot would trim out directionally prior to putting the gear down; however, the trim system would not be exactly centered at this time. When the gear was lowered, this minor out-of-trim condition was amplified by a ratio of four to one. Due to turbulence and low speed flight characteristics, this out-of-trim condition would go undetected until after landing when the rudders were released and the nosewheel steering engaged which resulted in an abrupt transient in the steering system. It was recommended that consideration be given to reducing the directional trim rate and incorporating a rudder position indicator. Cooper Rating of 3.0.

The compatibility of the roll and pitch force gradients was satisfactory; however, due to the large differences in the moments of inertia, the airplane responded much quicker in roll than in pitch or yaw. This was not considered to be a discrepancy against the airplane, but a characteristic of a very long and narrow configuration. It required some getting use to by the pilots in order not to overcontrol in roll, particularly with the tips up. A change should not be made in the response characteristics, but something should be done to reduce the adverse yaw due to ailerons. This would eliminate the primary objectionable characteristics in the airplane. If the pilots were not concerned about generating yaw with aileron inputs, the fighter-like roll response would not be objectionable. Control force compatibility: Cooper Rating of 3.0.

The trim change while operating the landing gear was negligible. The trim change while operating the wing tips was small and occurred at such a slow rate that it was hardly noticeable to the pilot since he took care of it in the normal trimming of the airplane. The trim change due to operation of the flaps was large, but easily manageable with the trim system. Although the trim system did take care of the trim change, when the flaps were lowered the control column moved very near the forward end of its travel leaving very little usable down elevon for maneuvering, go-around, or flying in turbulent air. This could be extremely hazardous in the instance where the center of gravity was farther aft than normal.

It was recommended that a design change be made whereby more down elevon control would be available with flaps down.



FLIGHT RATE PERFORMANCE

# FLIGHT HOUR PERFORMANCE



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# **TURN AROUND PERFORMANCE**

Turnaround Performance

WBS 4.41





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# DATA PARAMETER SUMMARY

TOTAL PARAMETER LIST	837
PARAMETERS DELETED	-249
PARAMETERS ADDED	+378
TOTAL PARAMETER LIST	<b>9</b> 66
PARAMETER DISTRIBUTION	
MISCELLANEOUS	138
ACCELEROMETERS	67
POSITIONS	104
TEMPERATURES	142
STRAIN GAGES	118
PRESSURES	397

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# ORIGINAL PAGE BLACK AND WHITE PHOTOGRAPH





# ORIGINAL PAGE BLACK AND WHITE PHOTOGRAPH



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COOPER PILOT OPINION RATING SYSTEM				
Adjective rating	Numerical rating	Description	Primary mission accomplished	Can be landed
	I	Excellent, includes optimum	Yes	Yes
Satisfactory	2	Good, pleasant to fly	Yes	Yes
	3	Satisfactory, but with some mildly unpleasant characteristics	Yes	Yes
	4	Acceptable, but with unpleasant characteristics	Yes	Yes
Unsatisfactory	5	Unacceptable for normal operation	Doubtful	Yes
	6	Acceptable for emergency condition only: Failure of stability augmenter	Doubtful	Yes
	7	Unacceptable even for emergency condition: Failure of stability augmenter	No	Doubtful
Unacceptable	8	Unacceptable—dangerous	No	No
	9	Unacceptable—uncontrollable	Νο	No
Catastrophic	10	Motions possibly violent enough to prevent pilot escape	No	No

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### PILOT'S FLIGHT REPORT

AIR VEHICLE	XB-70A #2 (207)	PILOT A. S. White
FLIGHT NO	(2-39)	COPILOT Col. J. Cotton
DATE OF TEST	19 May 1966	FLIGHT TIME 1:59
CHASE AIRCRAFT	AND PILOTS	
Chase 1 Chase 2 Chase 3 Support Rescue	$\begin{array}{rrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrr$	apt. Hoag/DeLong(NAA Photo) apt. Livingston apt. Hoag/DeLong(NAA Photo) t. Col. Fulton/Prahl q. Ldr. Cretney/Maj.Doryland

### PURPOSE OF FLIGHT

Mach 3 for 30 minutes!

# 1.0 PREFLIGHT AND TAXI

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The preflight and engine start phases were completed by 0814. It was necessary to make a soft start on the #l engine. Although this was not written up as a discrepancy, soft starts have been necessary for the last three engine starts.

An oil leak was detected from the engine 4 pod vent. No. 4 engine was accelerated to 90 per cent rpm for a short time and on reducing power to idle, the leak had stopped. No further action was taken.

Data was taken for the power advances required during the taxi phase. No brake chatter was encountered at any time on this flight.

#### 2.0 TAKEOFF, CLIMB, AND CRUISE

Brakes were released for takeoff at 0900 and MAX A/B was immediately selected. OVERSPEED was not used for this takeoff. The runway temperature was reported to be  $69^{\circ}$  at brake release. The acceleration check was made using the sensitive airspeed indicator on the pilot's side. According to the handbook, the airplane should have accelerated from 70 to 148 knots in 20 seconds. It actually accelerated from 70 knots to 150 knots in 20 seconds. Rotation started at 195 knots, and the airplane flew off the ground at 210 knots on the pilot's tape airspeed indicator.

The gear and flaps were raised. The #2 utility augmentation channel disengaged during the gear cycle and was immediately re-engaged after the gear was up and locked. The tips were lowered to one-half at 300 knots during the climb. At .9 Mn the AICS was reset and switched

to the AUTO mode. The first turn was made at 32,000 ft. fifty miles east of the Edwards TACAN. The airplane did not accelerate during the turn; consequently, it did not go supersonic until the turn was completed. The windshield was raised for the acceleration and the airplane was accelerated at 32,000 ft. to 575 knots where the climb was continued. During the initial part of the climb, the speed was allowed to build up to 595 knots and then slowly bled off to 575 by 40,000 ft.

The inlets started between 2.13 and 2.16 Mn just prior to reaching 50,000 ft. in the climb. After accelerating to 2.6 Mn, light buffeting was observed in the airplane of the type encountered on previous flights when the bleed air holes were partially plugged. However, this buffeting all but disappeared by 2.7 Mn and was not noted again during the flight.

Both the pilot and copilot had trouble in the climb phase of this flight with the face plate fogging over. The pilot normally runs the heater rheostat at the mid position, but had it on full hot and was still getting some fogging. The copilot reduced the airflow through the eyeball outlets to a minimum, and then turned the cabin heat up one notch in an attempt to help this situation. Later in the flight when the temperature was up, neither pilot had trouble with the face plate fogging. Whatever the cause, it appears that the face plate heating units are not adequate to take care of the moisture expelled in the helmet.

The turn north of Lovelock, Nevada, was made at approximately 2.6 Mn, and up until that time the TACAN indications were very good. The DME function of the TACAN broke lock just before the turn, so the ground station advised the pilot of the turning point. After rolling out of the turn, the TACAN locked on Lake View TACAN station and appeared to be functioning satisfactorily, but the heading information drifted and fooled the pilot into thinking that he was left of course when actually the radar track showed him to be right of course. The TACAN worked very well for the remainder of the flight. Any errors in navigation were caused by the drifting and the sluggishness of the heading information.

The inlet system was switched to HIGH PERFORMANCE at 2.7 Mn. The turn northeast of Boise, Idaho, was made at 2.8 Mn using a 15- to 17-degree bank angle. The airplane did not accelerate in this turn.

During the run between a point north of DuBois to Rock Springs, the airplane was slowly accelerating, but the throat Mns were on the lower limit. At 2.9 Mn the deviation control was set to 000 for the right inlet. At 2.92 the setting was changed to 995. At 2.97 the left inlet deviation control was set to 000. Mach 3 indicated was achieved at 0959 at a point in a turn just southeast of Rock Springs, Wyoming. Total fuel remaining at that time was 81,000 pounds. The throat Mns indicated 3.06 on the left and 3.02 on the right. Consequently, the

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right deviation was changed to 990. During the next two minutes, the Mach number built up to 3.025, and then bled off to 3.01 as the airplane climbed above 71,000 ft. Throat Mns were 3.05 on the left and 3.04 on the right. The deviation settings were then changed to 995 on the left and 985 on the right and remained that way for the rest of the Mach 3 portion of the flight. The Mach number varied between 3.02 and 3.06 with several constant periods at 3.03. Altitude variations ranged between 70,500 and 72,500 ft. It is the pilot's opinion that the altitude variations were caused by very minor errors in pitch attitude, rather than any abrupt pressure changes in the atmosphere. The altitude changes were not sudden; however, the rate-of-climb instrument showed some rather large excursions occasionally during the flight.

A right turn was started at Prescott, Arizona, and the airplane was held in a 15-degree bank. The flight path at the beginning of the turn was approximately ten miles west of the planned flight path, and the airplane overshot the planned flight path by approximately fifty miles on arriving in the Edwards area. No problem was encountered in maintaining speed in the turn, although altitude control required intense pilot concentration.

The throttle settings during the Mach 3 run varied between a 92-degree throttle angle and MAX A/B. During the early part of the run, the average throttle setting was approximately ninety-eight degrees. During the final ten minutes, the average throttle setting was 95 degrees or less.

The pilot was aware of the minor trim changes caused by fuel burning out of tank 8 and tank 1. In other words, the pilot could sense the point where tank 8 stopped feeding and tank 1 began feeding due to the shift in the CG. This has a minor effect on the precise altitude control, since only minor pitch trim changes will cause rather large changes in the rate of climb.

A marker was placed on the data at C/N 5688 with a call-out for correlation with the ground stations at Edwards during the last five minutes of the run. After at least 32 minutes at Mach 3, the inlets were switched to normal and the deviation controls set at 005. Before the pilot could reduce the throttle to Military Power, the left inlet unstarted. It restarted immediately and the copilot reset the duct. At approximately this time the right inlet unstarted and went into several cycles of buzz. The copilot selected LOW PERFORMANCE; but even though the airplane was buffeting moderately due to the inlet settings, the buzz lights went out and the side-to-side oscillation associated with buzz ceased. The VIBRATION HIGH light came on with no indication of high vibration on any of the twelve pickups. Military Power was selected and the deceleration commenced. As the airplane decelerated through 2.6 Mn, the left-hand #1 buzz light came on with no airplane transient. This occurred several times during the descent to Mach 2.

At approximately 2.2 Mn, in a right turn near Bishop, the windshield was lowered. Total fuel at this time was 26,500 pounds. At a speed just below Mach 2, the #6 engine EGT suddenly increased to 1000 degrees. The throttle was reduced to IDLE, eliminating the over-temperature condition. At .7 Mn it was noted that the utility pump status lights were on for utility pumps 1, 3, 4, and 5. At approximately 1.2 Mn, the #1 engine nozzle position indicator became inoperative and began to spin counterclockwise. Just after going subsonic with engines #1, #3, #4, and #6 at IDLE, the #3 engine compressor vibration pickup was indicating 60 per cent. The throttle was increased to 80 per cent rpm, which reduced the vibration to 40 per cent. The #2 engine was then reduced to IDLE for the descent.

### 3.0 APPROACH AND LANDING

The normal descent was made to a straight-in final approach. While turning on final approach after having lowered the gear satisfactorily, the flaps were lowered. The flaps came down and then slowly started back up again with the flap switch in the DOWN position. Flaps were raised on the approach and a no-flap landing was made. Touchdown occurred at approximately 174 knots on the pilot's VSI. Two chutes inflated and braking was satisfactory. After turning off the runway, during the Military Power runs, it was noted that the #4 nozzle had failed open. The #4 engine EGT indicated only 500 degrees at Military Power throttle setting, and the nozzle position was 70 per cent.

# 4.0 SUMMARY

This flight should have removed all doubt about the XB-70A's capability of accomplishing its objectives. Although some difficulty was encountered in accelerating at the northern end of the course, once Mach 3 was achieved it was no problem maintaining the speed, even in the turns.

It was apparent from this flight that the automatic inlet system has not been fully developed. Evidently more data at Mach 3 is required to optimize this system so that the copilot isn't required to manually adjust the automatic schedule so many times in order to satisfactorily complete a mission. Since this is one of the most important, if not the most important, new system in the airplane, it would be very beneficial to schedule sufficient flying at Mach 3 to overcome this deficiency; in fact, this is strongly recommended.

### 5.0 DISCREPANCIES

- 1. Yaw augmentation #2 disengaged several times during flight.
- 2. Pilot and copilot face plate heat inadequate.
- 3. #6 engine went overtemp. at Mil. Power during descent.
- 4. #1 engine nozzle indicator spinning during last portion of flight.

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- Flaps went down and then slowly bled up on final approach.
  #4 engine nozzle stuck open on Mil. Power run after landing.
  VIBRATION HIGH light on after unstart. Individual readouts were 0.K.

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A. S. White Aircraft Commander

## HANDLING QUALITIES QUESTIONNAIRE

FLIGHT 2-39 PILOT A. White	CONFIGURATION	Tips - Down; Gear and F	laps - Up
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LONGITUDINAL	MACH/ALT.	RATING	COMMENTS
TRIMMABILITY - Ability to hold: Airspeed Altitude Attitude	3.0; 70к	FACS ON 4	Attitude control, and consequently altitude control, require more than normal pilot concentration.
		Facs off	
MANEUVERABILITY - Ability to change: Airspeed Altitude Load Factor		FACS ON 4	
		FACS OFF	
8 RESPONSE TO TURBULENCE		. 3	
RESPONSE TO CONFIGURATION CHANGES			
OVERALL LONGITUDINAL RATING	>	4	
# HANDLING QUALITIES QUESTIONNAIRE

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PAGE 2

	MACH/ALT.	RATING		COMMENTS
TRIMMABILITY -		FACS ON	3	Holding a heading is not difficult, but selecting a new heading is difficult due to the lag or drift in the
Ability to hold: Heading Bank Angle		FACS OFF		heading information. The resentation was available.
MANEIVERABILITY -		FACS ON	4	
Ability to change. Heading Bank Angle		FACS OFF		
RESPONSE TO TURBULENCE			3	
OVERALL LATERAL-DIRECTIONAL RATING			3.5	
CONTROL HARMONY			3	





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# FOLDOUT FRAME 2

XB-70A Flight History Summary





Air Vehicle 1 Flight Statistics

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Flight		Tim	e	Altitudo Tomponoturo		Smood	Supersonic		Over Mach 2		Mach 3	
No.	Date	Flight	Cum	(ft)	(°F)	Mach No.	Flight	Cum	Flight	Cum	Flight	Cum
1	9-21-64	1:07	1:07	16,000	_	0.50	-	-	÷		-	-
2	10-5-64	:55	2:02	28,000	_	0.85	· _	-	-	-	-	-
3	10-12-64	1:35	3:37	35,400	-	1.11	:15	:15	-	-		-
4	10-24-64	1:25	5:02	46,300	81	1.42	:40	:55	-	-	-	-
5	2-16-65	1:10	6:12	45,000	118	1.60	:40	1:35	-	-	-	-
6	2-25-65	:53	7:05	35,000	<b>–</b>	. 97	-	1:35	-	-	-	-
7	3-4-65	1:37	8:42	50,200	167	1.85	1:00	2:35		-	-	-
8	3-24-65	1:40	10:22	56,100	236	2.14	1:14	3:49	:40	:40	. –	-
9	4-2-65	:54	11:16	34,500	-	0.95	-	3:49	-	:40	-	-
10	4-20-65	1:42	12:58	58,500	314	2.30	1:14	5:03	:50	1:30	-	-
11	4-28-65	1:43	14:41	64,300	348	2.45	1:16	6:19	:57	2:27		
12	5-7-65	1:25	16:06	65,000	425	2.60	:58	7:17	:33	3:00	-	-
13	6-16-65	1:37	17:43	65,000	429	2.60	1:15	8:32	:50	3:50	-	-
14	7-1-65	1:44	19:27	68,000	508	2.85	1:19	9:51	:50	4:40	-	-
15	7-27-65	1:43	21:10	66,000	514	2.82	1:05	10:56	:43	5:23	-	-
16	9-22-65	1:57	23:07	67,000	511	2.83	1:17	12:13	:56	6:19	-	_
17	10-14-65	1:47	24:54	70,000	586	3+	1:10	13:23	:46	7:05	:02	:02
18	11-4-65	2:04	26:58	46,000	170	1.86	1:17	14:40	-	7:05	-	:02
19	11-8-65	2:23	29:21	45,500	170	1.89	:46	15:26	-	7:05	-	:02
20	11-12-65	2:25	31:46	46,000	150	1.84	:36	16:02	-	7:05	-	:02
21	11-18-65	2:02	33:48	47,000	185	1.88	:50	16:52	-	7:05	-	:02
22	11-30-65	1:59	35:47	56,000	320	2.34	1:12	18:04	:53	7:58	-	:02
23	12-2-65	1:51	37:38	60,000	328	2.46	1:22	19:26	:34	8:32	-	:02
24	12-7-65	2:26	40:04	62,000	-	2.45	:40	20:06	:22	8:54	-	:02
25	12-10-65	2:18	42:22	50,700	-	1.82	:55	21:01	-	8:54	_	:02
26	12-14-65	2:10	44:32	20,000	-	0.95	-	21:01	- 1	8:54	-	:02
27	12-20-65	1:58	46:30	42,000	-	1.78	1:11	22:12	-	8:54	-	:02

**II-3**05

SD 72-SH-0003

WBS 4.41

		Tim	ie	A14.4 1		C 1	Super	sonic	Over M	lach 2	Macł	n 3
Plight No.	Date	Flight	Cum	(ft)	(°F)	Mach No.	Flight	Cum	Flight	Cum	Flight	Cum
28	12-22-65	2:35	49:05	34,000	-	1.42	:14	22:26	-	8:54	-	:02
29	1-3-66	2:41	51:46	26,000		0.94	-	22:26	-	8:54	-	:02
30	1-6-66	3:40	55:26	33,000	_	0.94	-	22:26	-	8:54	-	:02
31	1-11-66	1:35	57:01	46,000	. –	1.85	:46	23:12	-	8:54	-	:02
32	1-11-66	:58	57:59	27,000	-	0.95	-	23:12	-	8:54	-	:02
33	1-15-66	1:27	59:26	47,000	-	1.85	:50	24:02	-	8:54	-	:02
34	2-26-66	2:22	61:48	20,000	-	0.92	-	24:02	-	8:54	-	:02
35	3-3-66	2:42	64:30	15,000	-	0.55	-	24:02	-	8:54	-	:02
36	3-4-66	2:27	66:57	56,000	225	2.02	:48	24:50	:12	9:06	-	:02
37	3-7-66	2:19	69:16	67,000	290	2.22	1:02	25:52	:16	9:22	-	:02
38	3-22-66	2:11	71:27	32,000	-	0.97	-	25.52	-	9:22	-	:02
39	3-24-66	2:00	73:27	60,000	360	2.42	1:06	27:58	:21	9:43	-	:02
40	3-28-66	1:41	75:08	65,000	365	2.43	1:21	28:19	:32	10:15	-	:02
41	4-1-66	2:09	77:17	58,800	365	2.45	1:15	29:34	:20	10:35	-	:02
42	4-5-66	2:01	79:18	61,000	365	2.43	1:12	30:46	:48	11:23	-	:02
43	4-13-66	2:03	81:21	62,500	418	2.60	1:05	31:51	:47	12:10	-	:02
44	4-19-66	2:12	83:33	17,000	-	0.58	-	31:51	-	12:10	-	:02
45	4-21-66	2:02	85:35	61,000	360	2.42	1:20	33:11	:59	13:09	-	:02
46	4-25-66	2:07	87:42	63,000	415	2.55	1:12	34:23	:52	14:01	-	:02
47	4-27-66	2:41	90:23	31,000	-	1.50	:08	34:31	-	14:01	-	:02
48	5-3-66	1:22	91:45	23,000	· · · · · · · · · · · · · · · · · · ·	0.55		34:31		14:01	-	:02
49	5-9-66	2:16	94:01	15,000	-	0.50	-	34:31	-	14:01	-	:02
50	11-3-66	2:00	96:01	61,000	235	2.10	:35	35:06	:12	14:13	-	:02
51	11-10-66	1:39	97:40	60,000	385	2.50	1:12	36:18	:32	14:45	-	:02
52	11-23-66	1:38	99:18	61,000	395	2.51	1:06	37:24	:22	15:07	-	:02
53	12-12-66	1:57	101:15	60,000	357	2.52	1:01	38:25	:32	15:39	-	:02
54	12-16-66	1:54	103:09	60,300	370	2.55	1:01	39:26	:27	16:06	-	:02

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Air Vehicle 1 Flight Statistics (Cont)

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		Ti	me		-	C 1	Supers	sonic	Over M	lach 2	Mach	n 3
Flight No.	Date	Flight	Cum	Attitude (ft)	(°F)	Speed Mach No.	Flight	Cum	Flight	Cum	Flight	Cum
55	12-20-66	1:45	104:54	60,800	380	2.53	1:00	40:26	:32	16:38	-	:02
56	1-4-67	1:44	106:38	60,400	385	2.53	:57	41:23	:35	17:13	-	:02
57	1-13-67	1:46	108:24	61,000	392	2.57	1:00	42:23	:38	17:51	-	:02
58	1-17-67	1:44	110:08	60,200	403	2.54	1:01	43:24	:38	18:29	-	:02
59	1-25-67	1:32	111:40	35,000	-	1.41	:51	44:15	-	18:29	-	:02
60	1-31-67	1:32	113:12	37,000	-	1.40	:49	45:04		18:29	-	:02
61	4-25-67	1:07	114:19	17,000	-		-	45:04	-	18:29	-	:02
62	5-12-67	2:18	116:37	16,500	-	-	-	45:04	-	18:29	-	:02
63	6-2-67	2:23	119:00	42,000	-	1.43	:42	45:46	-	18:29	-	:02
64	6-22-67	1:54	120:54	54,000	179	1.83	:57	46:43	-	18:29	-	:02
65	8-10-67	2:29	123:23	15,500	-	0.92	-	46:43	-	18:29	-	:02
66	8-24-67	1:52	125:17	58,000	-	2.27	:56	47:39	:23	18:52	-	:02
67	9-8-67	1:55	127:10	59,700	-	2.30	1:15	48:54	:51	19:43	-	:02
68	10-11-67	1:39	128:49	58,000	-	2.43	1:11	50:05	:47	20:30	-	:02
69	11-2-67	1:56	130:45	64,000	403	2.55	1:14	51:19	:47	21:17	-	:02
70	1-12-68	1:54	132:39	67,000	-	2.55	1:01	52:20	:42	21:59	-	:02
71	2-13-68	2:43	135:22	41,000	-	1.18	:16	52:36	-	21:59	-	:02
72	2-28-68	1:51	137:13	18,500		-	-	52:36	-	21:59	-	:02
73	3-21-68	2:32	139:45	15,500	-	-	-	52:36	-	21:59	-	:02
74	6-11-68	1:11	140:56	9,500	-	-	-	52:36	-	21:59	- 1	:02
75	6-28-68	2:39	143:35	39,400	-	1.23	:18	52:54	-	21:59	-	:02
76	7-19-68	1:55	145:30	42,000	-	1.62	:48	53:42	-	21:59	-	:02
77	8-16-68	1:55	147:25	63,000	-	2.47	:55	54:37	:34	22:33	-	:02
78	9-10-68	1:48	149:13	63,000	-	2.54	1:04	55:41	:35	23:08	-	:02
79	10-18-68	1:56	151:09	52,000	-	2.18	:54	56:35	:13	23:21	-	:02
80	11-1-68	2:08	153:17	41,000	-	1.62	:48	57:23	-	23:21	-	:02
81	12-3-68	1:58	155:15	39,400	-	1.64	:56	58:19	-	23:21	-	:02
82	12-17-68	1:45	157:00	63,500	-	2.53	1:12	59:31	:55	24:16	-	:02
83	2-4-69	3:16	160:16	29,000	-	0.92	-	59:31	-	24:16	-	:02

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# Air Vehicle 2 Flight Statistics

# WBS 4.41

		Tim	ıe				Supers	sonic	Over M	1ach 2	Mac	h 3
Flight No.	Date	Flight	Cum	Altitude (ft)	Temperature (°F)	Speed Mach No.	Flight	Cum	Flight	Cum	Flight	Cum
1	7-17-65	1:13	1:13	42.000	80	1.41	:21	:21	-	-	_	-
2	8-10-65	1:27	2:40	41,000	81	1.45	:31	:52	-	-	-	-
3	8-18-65	1:58	4:38	46,000	80	1.45	:44	1:36	- 1	-	-	-
4	8-20-65	2:04	6:42	42,000	81	1.44	:41	2:17	<u> </u>	-	-	-
5	9-17-65	1:55	8:37	50,500	165	1.83	1:00	3:17	-	-	-	-
6	9-29-65	1:44	10:21	54,000	295	2.23	:32	3:49	:05	:05	-	-
7	10-5-65	1:40	12:01	55,000	290	2.30	:31	4:20	:09	:14	-	
8	10-11-65	1:55	13:56	57,500	305	2.34	1:18	5:38	:53	1:07	· -	-
9	10-16-65	1:43	15:39	59,500	360	2.43	1:12	6:50	:47	1:54	-	-
10	10-20-65	2:07	17:45	59,000	326	2.46	1:11	8:01	:43	2:37	-	. –
11	11-2-65	1:54	19:40	59,000	330	2.45	1:20	9:21	:46	3:23	-	-
12	11-29-65	2:19	21:59	15,200	-	. 53	-	9:21	-	3:23	-	-
13	12-1-65	2:02	24:01	64,000	435	2.67	1:24	10:45	:59	4:22		-
14	12-3-65	1:55	25:56	69,000	510	2.87	1:14	11:59	:51	5:13	-	
15	12-11-65	2:03	27:59	70,600	556	2.94	1:16	13:15	:53	6:06	-	- 1
16	12-21-65	1:49	29:48	72,000	535	2.95	1:20	14:35	1:02	7:08	-	-
17	1-3-66	1:52	31:40	72,000	601	3.05	1:22	15:57	1:06	8:14	:03	:03
18	1-12-66	1:46	33:28	72,000	605	3.06	1:24	17:21	1:03	9:17	:04	:07
19	2-7-66	2:11	35:39	42,000	-	1.44	1:03	18:24	-	9:17	-	:07
20	2-9-66	1:49	37:28	70,800	606	3.04	1:09	19:33	:48	10:05	:05	:12
21	2-16-66	3:06	40:34	32,000		1.10	:02	19:35	-	10:05	-	:12
22	2-17-66	1:47	42:21	73,000	608	3.04	1:23	20:58	1:05	11:10	:15	:27
23	3-10-66	1:56	44:17	67,000	445	2.76	1:25	22:23	:54	12:04	· -	:27
24	3-15-66	1:59	46:16	69,500	508	2.85	1:33	23:56	1:11	13:15	-	:27
25	3-17-66	1:52	48:08	70,350	508	2.85	1:27	25:23	1:09	14:24	-	:27
26	3-19-66	1:57	50:05	74,000	550	2.93	1:30	26:53	1:10	15:34	-	:27

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Air Vehicle 2 Flight Statistics (Cont)

	W	BS	4.	41
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Flight		Tin	ne	Altitude Temperature		Speed	Supersonic		Over Mach 2		Mach 3	
No.	Date	Flight	Cum	(ft)	(°F)	Mach No.	Flight	Cum	Flight	Cum	Flight	Cum
27	3-24-66	1:32	51.37	64,000	440	2.71	:47	27:40	:29	16:03	-	:27
28	3-26-66	3:09	54:46	36,000	-	. 94	-	27:40	-	16:03	-	:27
29	3-29-66	1:51	56:37	48,000	120	1.65	1:25	29:05	-	16:03	-	:27
30	3-31-66	2:10	58:47	72,000	550	2.95	1:20	30:25	1:02	17:05	-	:27
31	4-4-66	1:57	60:44	73,000	550	2.95	1:30	31:55	1:09	18:14	-	:27
32	4-8-66	2:05	62:49	73,000	610	3.07	1:09	33:04	:50	19:04	:16	:43
33	4-12-66	1:49	64:38	72,800	624	3.08	1:17	34:21	:53	19:57	:20	1:03
34	4-16-66	2:01	66:39	71,000	568	3.03	I:03	35:24	:43	20:40	:01	1:04
35	4-23-66	2:01	68:40	66,000	473	2.73	1:22	36:46	1:01	21:41	-	1:04
36	4-26-66	2:05	70:45	65,500	423	2.65	1:17	38:03	:50	22:31	-	1:04
37	4-30-66	2:16	73:01	16,000	-	. 55	-	38:03	-	22:31	-	1:04
38	5-16-66	2:09	75:10	65,000	470	2.73	1:11	39:14	:46	23:17	-	1:04
39	5-19-66	<b>1:</b> 59	77:09	72,500	620	3.06	1:31	40:45	1:13	24:30	:33	1:37
40	5-22-66	2:22	79:31	36,500	110	1.51	:25	41:10	-	24:30	-	1:37
41	5-25-66	2:23	81:54	42,000	115	1.63	:49	41:59	-	24:30	-	1:37
42	5-27-66	2:08	84:02	62,000	401	2.53	1:17	42:16	:39	25:09	_	1:37
43	5-31-66	2:02	86:04	57,000	290	2.25	1:12	43:28	:22	25:31	_	1:37
44	6-4-66	2:05	88:09	70,000	555	2.93	1:20	44:48	:54	26:25	-	1:37
45	6-6-66	2:00	90:00	72,000	620	3.05	1:15	-	:53	27:18	:09	1:46
46	6-8-66	2:13	92:22	32,000	85	1.41	:14	47:17	-	27:18	-	1.46

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WBS CODE: 4.41

#### COST DEFINITION:

The cost data presented for this WBS level 5 item (WBS 4.41 Flight Test) is a summation of the level 6 blocks as indicated below:

			Cost Data
WBS 4.41	Flight Test	\$ 2,781,972	Page II-315
WBS 4.41.1	Flight Operations	\$29,594,412	Page II-385
WBS 4.41.2	Air Vehicle Maintenance	\$ 5,216,014	Page II-432
WBS 4.41.3	Instrumentation	\$ 1,164,316	Page II-448
WBS 4.41.4	Post Flight Evaluation	\$ -0-	
WBS 4.41.5	Major Air Vehicle Repair	\$ 496,336	Page II-470
		\$39,253,050	Page II-341

Detail cost data can be found on the pages indicated above.

WBS 4.41 Flight Test

Cost data accumulated in this WBS item are those expenditures appearing in the Cost Accumulation Statements that can not be associated with a Flight Test level 6 item. The costs include engineering, flight line production support, and test technicians.

#### WBS 4.41.1 Flight Operations

Cost accumulated in this level 6 item contain all identifiable expenditures associated with the planning, ground operations and flight operations that existed during the XB-70 Flight Test Development Program. The technical discussion on page II-379 provides a detail discussion of the activities associated with this WBS item.

#### WBS 4.41.2 Air Vehicle Maintenance

\$ 5,216,014

\$ 2.781.972

\$29.594.412

This WBS item contains those costs associated with the scheduled and unscheduled maintenance of the two aircraft during the flight test portion of the XB-70 program. Major repairs as the result of accidents or malfunctions **are not** included in these costs. Flight Test Ground Support Equipment maintenance is likewise not included (see WBS 5.51). The technical discussion beginning on page II-410 provides a detail discussion of the maintenance concept and activities associated with the flight test program.



WBS CODE: 4.41

#### WBS 4.41.3 Instrumentation

\$ 1,164,316

Expenditures associated with the calibration, maintenance, data acquisition, and reliability of the flight test instrumentation are collected in this WBS item. Costs associated with the development and installation of the test instrumentation is contained in WBS 1.11 Test Instrumentation Subsystem. The technical discussion on page II-438 provides a detail discussion of the flight test instrumentation.

#### WBS 4.41.4 Post Flight Evaluation

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No cost data has been assigned to this item as the recorded cost records did not provide detail in this area. The evaluation of the flight test data is contained within the subsystem engineering effort as the design and support groups provided personnel to perform this function.

#### WBS 4.41.5 Major Air Vehicle Repair

496,336

Flight Test personnel supporting the repair of ship #1 after flights 1. 12 and 14 and ship #2 after flight 1 are contained in this item. Support from manufacturing for these repairs are included in WBS 1.12 as they can not be identified. Normal maintenance and major rework not the result of accidents are included in WBS 4.41.2 Air Vehicle Maintenance.

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

### 4-SYSTEM 4 FLIGHT TEST

5-SUBSYS 41 TOT HOURS HOURS DOLLARS DOLL DESIGN/ENGINEERING 2279153 2279153 LABOR AT \$ 5.601 12765763 12765 ENGR BURDEN AT \$ 6.677 15219035 15219	ARS 763 035
41 TOTA HOURS HOURS DOLLARS DOLL DESIGN/ENGINEERING 2279153 2279153 LABOR AT \$ 5.601 12765763 12765 ENGR BURDEN AT \$ 6.677 15219035 15219	ARS 763 035
HOURS HOURS DOLLARS DOLL DESIGN/ENGINEERING 2279153 2279153 LABOR AT \$ 5.601 12765763 12765 ENGR BURDEN AT \$ 6.677 15219035 15219	ARS 763 035
DOLLARS DOLL DESIGN/ENGINEERING 2279153 2279153 LABOR AT \$ 5.601 12765763 12765 ENGR BURDEN AT \$ 6.677 15219035 15219	ARS 763 035
DESIGN/ENGINEERING 2279153 2279153   LABOR AT \$ 5.601 12765763 12765   ENGR BURDEN AT \$ 6.677 15219035 15219	763 035
DESIGN/ENGINEERING 2279153 2279153   LABOR AT \$ 5.601 12765763 12765   ENGR BURDEN AT \$ 6.677 15219035 15219	763 035
LABUR AT \$ 5.601 12765763 12765   ENGR BURDEN AT \$ 6.677 15219035 15219	763 035
ENGR BURDEN AT \$ 6.677 15219035 15219	035
PRODUCTION 11030 11030	
LABOR AT \$ 3.225 35575 35	575
SHOP SUPPORT 301586 301586	
LABOR AT \$ 4,042 1219027 1210	0 27
PLANN ING 77 77	021
LABOR AT \$ 3,883 200	200
IEST/00 274000 274000	2.33
	っっと
MEC RIDDEN AT \$ F 900 (05100) (051	227
HEG BUKDEN AT \$ 5.899 4051204 4051	204
ENGR MATERIAL 1577071 1577	071
MFG MATERIAL 144913 144	913
MPC 276599 276	599
OTHER COST 1202278 1202	278
SUB-TOTAL 38237099 38237	099
GEN & ADMIN 1015951 1015	951
TOTAL COST 39253050 39253	<b>dan</b> 1999

COST DETAIL FOR 4.41 TOTALS ON PAGE II-341

> COST DETAIL FOR WBS LEVEL 6 ITEMS INCLUDED IN 4.41 SEE PAGE **II-313** ·

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

#### 4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT TEST

			6-M ASSY	6-M ASSY	6-M ASSY	6-M ASSY
				HUIRS		
				DOLLARS	00005	DOLLARS
			DOLLAND		WOLLANG	DOLLAND
DESIGN/ENGINEER	ING		7882	1958503	219064	84141
LABOR AT \$	5.601		31041	11308624	953328	420445
ENGR BURDEN	AT \$	6.677	35715	13498804	1167125	459205
PRODUCTION			11030			
LABOR AT \$	3.225		35575			
SHOP SUPPORT			28298	81949	148211	9628
LABOR AT \$	4.042		96006	304273	658064	33160
PLANN ING			77			
LABOR AT \$	3.883		299			
TEST/QC			179735	122675	64799	578
LABOR AT S	4.665		828056	587413	299264	2217
MFG BURDEN	AT \$	5.899	1367107	1223461	1224034	4720 <b>7</b>
ENGR MATERIAL			277387	430206	698057	149948
MFG MATERIAL				144913		
MPC			<b>43</b> 566	99768	102548	24624
OTHER COST				1191690	10528	
SUB-TOTAL			2714752	28789152	5112948	1136806
GEN & ADMIN			67220	805260	103066	27510
TOTAL COST			2781972	29594412	5216014	1164316

SUBDIVISION OF WORK				
COST DETAIL - SEE PAGE	II <b>-</b> 315	<b>II-</b> 385	<b>II-</b> 4 <b>3</b> 2	<b>II-</b> 448

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 FL IGHT TEST

		6-M ASSY 05 HOURS DOLLARS	TOTAL Hours Dollars
DESIGN/ENGINEERING		9563 2	2279153 /
LABOR AT \$ 5.601		52325	12765763
ENGR BURDEN AT \$	6.677	58186	15219035
PRODUCTION			11030
LABOR AT \$ 3.225			35575
SHOP SUPPORT		33500	301586
LABOR AT \$ 4.042		127524	1219027
PLANN ING		н. -	77
LABOR AT \$ 3.883			2 99
TEST/QC		6312	374099
LABOR AT \$ 4.665		28385	1745335
MFG BURDEN AT \$	5.899	189395	4051204
ENGR MATERIAL		21473	1577071
MFG MATERIAL			144913
MPC		6093	276599
OTHER COST		60	1202278
SUB-TCTAL		483441	38237099
GEN & ADMIN		12895	1015951
TOTAL COST		496336	39253050

COST DETAIL		
SEE PAGE	<b>II-</b> 470	<b>II-</b> 341

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

FLIGHT TEST

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 0

		PROD HOURS DOLLARS	TEST /QC HOURS DOLLARS	TOTAL HOURS DOLLARS
DESIGN/ENGINEERING			7882	7882
LABOR AT \$ 3.938			31041	31041
ENGR BURDEN AT \$	4.531		35715	35715
PRODUCTION		11030		11030
LABOR AT \$ 3.225		355 <b>7</b> 5		35575
SHOP SUPPORT			28298	28298
LABOR AT \$ 3.393			960.06	96006
PLANN ING			77	77
LABOR AT \$ 3.883			299	299
TEST/QC	•	173005	6730	179735
LABOR AT \$ 4.607		803054	25002	828056
MFG BURDEN AT \$	6.239	1211502	155605	1367107
ENGR MATERIAL	- · · · ·		277387	277387
MPC		•	43566	43566
SUB-TOTAL		2050131	664621	2714752
GEN & ADMIN		53828	13392	67220
TOTAL COST		2103959	678013	2781972

TIME-PHASED COST			
DETAIL - SEE PAGE	<b>II-316</b>	II-320	<b>II-</b> 330

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# PRODUCTION

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 0 SUBD OF WORK PRODUCTION

# FLIGHT TEST

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-3	63					52259	52259
Q-4	63						
0-1	64					40112	40112
Q-2	64						10
Q-3	64	19.5	3486	3.201	11157	243 330	254487
Q-4	64					- 15 550	234401
Q-1	65	31.5	5515	3.224	17781	211916	220607
Q-2	65					ETT 110	227071
Q-3	65	12.0	2029	3.271	663 <b>7</b>	220272	226000
Q-4	65				00.51		220909
Q-1	66					271514	271514
Q-2	66					211714	211314
Q-3	66					172099	172099
тот	AL	63.0	11030		35575	1211502	1247077

#### TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

FLIGHT TEST

# TEST/QC

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 0 SUBD OF WORK PRODUCTION

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + Burden \$
Q-3	63 63	60.0	10129	3.481	35264		35264
Q-1 Q-2	64 64	39.0	6630	3.903	25878		25878
Q-3 Q-4	64 64	204.0	35787	4.275	153001		153001
Q-1 Q-2	65 65	162.0	28136	4.521	127199		127199
Q-3 Q-4	65 65	193.5	32492	4.946	160709		160709
Q-1 Q-2	<b>6</b> 6	208.5	36204	5.013	181473	• •	181473
Q-3	66	141.0	23627	5.059	119530		119530
тот	AL	1008.0	173005		803.054		803054

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 0 SUBD OF WORK PRODUCTION

#### FLIGHT TEST

	MAN- MONTHS	LABOR HOURS	LABOR Rate	LABOR	BUR DEN DOLL ARS	LABOR + BURDEN \$	GEA
0-3 63	60.0	10129	3.481	35264	52259	87523	1463
Q = 4 63 Q = 1 64 Q = 2 64	39.0	6630	3.903	25878	40112	<b>65 9</b> 90	1404
$Q=2 \ 64$ $Q=3 \ 64$	223.5	39273	4.180	164158	243330	407488	8671
Q = 1 65 Q = 2 65	193.5	33651	4.308	144980	211916	356896	9522
Q-3 65 Q-4 65	205.5	34521	4.848	167346	220272	387618	10342
Q - 1 66 Q - 2 66	208.5	36204	5.013	181473	271514	452987	13643
Q-3 66	141.0	23627	5.059	119530	172 099	291629	8783
TOTAL	1071.0	184035		838629	1211502	2050131	53828

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM45-SUBSYSTEM416-MAJASSY00SUBD0FWORKPRODUCTION

	TOTAL
	COST
0 7 / 7	00000
Q-3 03	88,886
Q-4 63	
Q-1 64	67394
Q-2 64	
Q-3 64	416159
Q-4 64	
Q-1 65	366418
Q-2 65	
Q-3 65	397960
Q-4 65	
Q-1 66	466630
Q-2 66	
0-3 66	300412
TOTAL	2103959

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING

4-SYSTEM	4	
5-SUB SYSTEM	41	FT.TCHIP PERSON
6-MAJ ASSY	0	THEORE INCL
SUBD CF WORK	TEST/QC	

		MAN- MONTHS	LABOR	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
0-1	58						
$\overline{0-2}$	58						
$\hat{Q} - \hat{3}$	58						
0-4	58						
Q-1	59						
Q-2	59						
0-3	59						
Q-4	59						
Q-1	60						
Q-2	60						
Q-3	60						
Q-4	60						
Q-1	61		•				
Q-2	61						
Q-3	61	1.5	302	3.460	1045	1182	2227
Q-4	61						
Q-1	62	4.5	784	4.061	3184	3623	6807
Q-2	62						
Q-3	62	4.5	729	4.255	3102	3748	6850
Q-4	62						
Q-1	63	7.5	1234	4.219	520 <b>6</b>	6506	11712
Q-2	63						
Q-3	63		52	6.904	359	<u>\$2</u>	451
Q-4	63	. · .					
0-1	64	3.0	549	3.776	2073	2870	4943
Q-2	64	•	<b>F a a</b>				
Q-3	64	3.0	582	4.784	2784	3718	6502
₩-4	64	1 5		2 02 <b>5</b>	10/0	1000	
Q-1	65	1.5	331	3.985	1343	1939	3282

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING 4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 0 SUBD OF WORK TEST/QC FLIGHT TEST

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-2 65						
Q-3 65	3.0	426	4.345	1851	1943	3794
Q-4 65						5121
Q-1 66	15.0	2545	3.495	8896	8897	17793
Q-2 66						
Q-3 66	1.5	342	3.503	1198	1197	2395
TOTAL	45.0	7882		31041	35715	66756

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP SUPPORT	
4-SYSTEM	4	
5-SUBSYSTEM	41	FLIGHT TEST
6-MAJ ASSY	0	
SUBD CF WORK	TEST/QC	

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	59						
Q-2	59	1					
0-3	59						
Q-4	59						
Q-1	60						
Q-2	<b>6</b> 0						
Q-3	<b>6</b> 0						
Q-4	60						
Q-1	61						
Q-2	61						
Q-3	61		2	2.500	5	7	12
Q-4	61						
Q-1	62	9.0	1596	3.073	4905	6048	10953
Q-2	62						
Q-3	62	16.5	2797	2.883	8.06 5	11573	19638
Q-4	62						
Q-1	63	18.0	3120	2.952	9209	16490	25699
Q-2	63						
Q-3	63	24.0	3969	3.353	13307	25623	38930
Q-4	63	20.0	5004				
0-1	54	30.0	5026	3.513	1/655	25156	42811
Q-2	04	21 6	5500	2 000		<b>A</b> ( <b>A</b> ) A	
0-4	04 44	51+5	9900	3.000	21382	30232	57614
0-1	04 45	14 6	2012	3 305	0000	1202/	2226
0-2	45	10.0	2912	2.202	2000	13,430	23209
W-2 0-2	65	10 5	3160	2 61 2	11415	10/50	20.94 F
0-4	65	1207	5100	5.012	11413	19400	20002
Q-1	66	1.5	210	3.343	702	1 086	1788

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT TEST 6-MAJ ASSY 0 SUBD OF WORK TEST/QC

	MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q-2 66 Q-3 66		6	4.667	28	4	32
TOTAL	166.5	28298	· ·	96006	155605	251611

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

FLIGHT TEST

PLANNING 4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 0 SUBD OF WORK TEST/QC

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	MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1 63 Q-2 63		3	2.333	. 7	10	17
Q-3 63		74	3.946	292	-10	282
TOTAL		77		299		299

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

FLIGHT TEST

# TEST/OC

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 0 SUBD OF WORK TEST/QC

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-1	62		70	2.929	205		205
Q-2	62						
Q-3	62		101	3.119	315		315
Q-4	62						
Q-1	63	6.0	943	3.131	2953		2052
Q-2	63						662
Q-3	63	13.5	2164	3.413	7386		7296
Q-4	63						1000
Q-1	64		120	3-808	457		457
Q-2	64		· · · ·				1.5
Q-3	64	12.0	2210	4.108	9079		0070
Q-4	64				2013		3013
0-1	65		43	4.744	204		20.4
Q-2	65						204
Q-3	65	6.0	1033	4.103	4238		4738
Q-4	65						7200
Q-1	66		45	3.711	167		167
0-2	66		••		101		107
Q-3	66		<b>1</b> .	1.999	-2	· · · · ·	-2
тот	AL	37.5	6730		25002		25002

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 0 SUBD OF WORK TEST/QC

# FLIGHT TEST

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
0-1	58							
Q-2	58							
0-3	58							
Q-4	58							
Q-1	59							
Q-2	59							
Q-3	59							
Q-4	59							
Q-1	60							
Q-2	60							
Q-3	60							
Q-4	60							
ର-1	61				r			
Q-2	61							
Q-3	61	1.5	304	3.454	1050	1189	2239	56
0-4	61		•.					2.0
Q-1	62	13.5	2450	3.385	8294	9671	17965	6191
Q-2	62							
Q-3	62	21.0	3627	3.166	11482	15321	26803	48918
Q-4	62							
Q-1	63	31.5	5300	3.278	17375	23006	40381	55915
Q-Z	63							
Q-3	63	37.5	6259	3.410	21344	25705	47049	44862
Q-4	63							
	64	33.0	5695	3.544	20185	28026	48211	55071
	64							
	04	40.5	8292	4.009	33245	39950	73195	54756
0.1	04 45	10.0	2000	2 225				
w~1 ∩?	00 45	10.U	3292	3.305	10880	15875	26755	11832
<u>a</u> -2	09 45	20 E	4710	2 700	1750			
Q-0	00	20.3	4019	3.190	17504	21 393	38897	-214

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 0 SUBD OF WORK TEST/QC

FLIGHT TEST

	MAN- MON TH S	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DULL ARS	LABOR + Burden \$	ENGR Matl
0-4 65							
Q-1 66	16.5	2800	3.488	9765	9983	19748	
Q-3 66	1.5	349	3.507	1224	1201	2425	
TOTAL	249.0	42987		152348	191320	343668	277387

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

FLIGHT TEST

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 0 SUBD CF WORK TEST/QC

		MPC	OTHER COST	SUB Total	G & A	TOTAL COST
Q-1 Q-2	58 58					
Q-3	58					
Q-4	58					
0-1	59					
Q-2	59					
Q-3	59					
Q-4	59					
Q-1	60					
Q-2	60					
Q-3	5U ( )					
Q-4	60		,			
0 2	01 (1					
0-2	61	r				
	01 41	5		2300	38	2338
0-1	61	4 3 0	24	<b>n</b> // <b>n</b> /		
0-2	62	400	54	24678	.414	25092
0-2	62	3955	1 6	70501	100/	<b>A A A A</b>
0-4	62		10	19091	1336	80927
<u>0</u> -1	63	5507		101002	1700	102505
0-2	63	5501		101003	1702	103505
$\overline{0}-\overline{3}$	63	4419	-49	96281	1610	07801
Q-4	63			10201	1010	21021
Q-1	64	5871		109153	2323	111476
Q-2	64	-			inn af San af	111410
C-3	64	19920		147871	3146	151017
0-4	64					
Q-1	65	3539		42126	1124	43250
Q-2	65					
Q-3	65	-38		38645	1031	39676

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

FLIGHT TEST

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 0 SUBD OF WORK TEST/QC

	MPC	OTHER COST	SUB TOTAL	G & A	TOTAL COST
Q-4 65		•			
0-1 65			19748	595	20343
Q-2 66					
Q-3 66			2425	73	2498
TOTAL	43566		664621	13392	678013

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESIGN/ENGINEER	ING
4-SYSTEM	4	
5-SUB SYSTEM	41	FLIGHT TEST
6-MAJ ASSY	0	

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58						
Q-2	58						
Q-3	58						
0-1	20						
n = 2	50						
0-3	59						
0-4	59						
Q-1	60						
Q-2	60						
Q-3	60						
Q-4	60						
Q-1	61						
Q-2	61						
Q-3	61	1.5	302	3.460	1045	1182	2227
Q = 4	61		704	<i>.</i>			
	02 62	4.5	184	4.061	3184	3623	6807
0-3	62	4.5	720	4 355	2102	1 2710	(
0-4	62		127	4.200	5102	3 148	6850
Q-1	63	7.5	1234	4,219	5206	6506	11712
Q-2	63				2200	0,000	11/12
Q-3	63		52	6.904	359	. 92	451
Q-4	63						121
Q-1	64	3.0	549	3.776	2073	2870	4943
Q-2	64						
Q-3	64	3.0	582	4.784	2784	3718	6502
Q-4	04 (E	1 6	~~ ~				
0-2	65	1.3	337	3.985	1343	1939	3282
w-2	0.5						

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING 4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT TEST 6-MAJ ASSY 0

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-3 65	3.0	426	4.345	1851	1943	3794
Q-4 65 Q-1 66	15.0	2545	3.495	8896	8 897	17793
Q-2 66 Q-3 66	1.5	342	3.503	1198	1197	2395
TOTAL	45.0	7882		31041	35715	66756

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

APRIL 1972

4-SYSTEM	PRUDUCTION 4	FLTCH	mrcm
6-MAJ ASSY	41 0		

		MAN- MON TH S	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLL ARS	LABOR + BURDEN \$
Q-3 Q-4	63 63					52259	52259
Q-1 Q-2	64 64					40112	40112
Q-3 Q-4	64 64	19.5	3486	3.201	11157	243330	254487
Q-1 Q-2	65 65	31.5	5515	3.224	17781	211916	229697
Q-3 Q-4	65 65	12.0	2029	3.271	6637	220272	226909
Q-1 Q-2	66 66					271514	271514
Q-3	66		•			172099	172099
TOT	AL	63.0	11030		35575	1211502	1247077

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP	SUPPORT		
4-SYSTEM	4	۰.	,	
5-SUB SYSTEM	41	× .	FLIGHT TEST	
6-MAJ ASSY	0			

		MAN- MON TH S	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1	59						
Q-2	59						
Q-3	59						
Q-4	59						
Q-1	60						
Q-2	60						•
Q-3	60						
Q-4	60	N.					
Q-1	61				<b>`</b> .		
Q-2	61					_	
Q-3	61		2	2.500	5	7	12
Q-4	61						10050
Q-1	62	9.0	1596	3.073	4905	6048	10953
Q-2	62	1 ( 5	2707	2 002	9945	11570	10/20
0-6	62	10.0	2191	2.883	8065	115/5	1,9038
0-1	62	18.0	3120	2 05 2	9200	16400	25600
0-2	63	10.0	5120	L • 7 J L	. 7207	10450	20077
0-3	63	24.0	3969	3, 353	13307	25623	38930
0-4	63	2100	3707	2000	10001		
0-1	64	30.0	5026	3,513	17655	25156	42811
Q-2	64						
Q-3	64	31.5	5500	3.888	21382	36232	57614
Q-4	64						
Q-1	65	16.5	2912	3.205	9333	13936	23269
Q-2	65						
Q-3	65	19.5	3160	3.612	11415	19450	30865
Q-4	65						
Q-1	66	1.5	210	3.343	702	1086	1788
Q-2	66						

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP	SUPPORT	
4-SYSTEM	4		
5-SUB SYSTEM	41		FLIGHT TEST
6-MAJ ASSY	0		

	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLAR S	BURDEN DOLLARS	LABOR + BURDEN \$
Q-3 66		6	4.667	28	4	32
TOTAL	166.5	28298		96006	155605	251611

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

FLIGHT TEST

	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1 63		3	2.333	7	10	17
Q-2 63 Q-3 63		74	3.946	292	-10	282
TOTAL		77		299		299
## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

O-MAJ ASST U	4-SYSTEM 5-SUBSYSTEM 6-MAJ ASSY	TEST/QC 4 41 0	FLIGHT TEST
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		MAN- MONTHS	LABOR	LABOR	LABOR DOLLARS	BUR DEN DOLL AR S	LABOR + BURDEN \$
Q-1	62		70	2.929	205		205
Q-2 Q-3	62 62		101	3.119	315		315
Q-4 Q-1	62 63	6.0	943	3.131	2953		2953
Q-2 Q-3	63 63	73.5	12293	3.469	42650		42650
Q-4 Q-1	63 64	39.0	6750	3.901	26335		26335
Q-2 Q-3	64 64	216.0	37997	4.266	162080		162080
Q-4	64 65	162.0	28179	4 521	127402		127402
Q-2	65 (5	100 5	20117	4.020	127403		121403
Q-4	65	199.5	39545	4.920	164947		164941
Q-1 Q-2	66 66	208.5	36249	5.011	181640		181640
Q-3	66	141.0	23628	5.059	119528		119528
TOT	AL	1045.5	179735		828056		828056

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

FLIGHT TEST

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 0

		MAN- MONTHS	L ABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL AR S	LABOR + Burden \$	ENG MAT
0-1	58							
0-2	58							
0-3	58							
0-4	58							
0-1	59							
Q-2	59							
$\overline{0-3}$	59							
0-4	59							
0-1	60							
Q-2	60		•					
Q-3	60							
Q-4	60							
Q-1	61							
Q-2	61							
Q-3	61	1.5	304	3.454	1050	1189	2239	5
Q-4	61							
Q-1	62	13.5	2450	3.385	8294	9671	17965	619
Q-2	62							
Q-3	62	21.0	3627	3.166	11482	15321	26803	4891
Q-4	62						(	
Q-1	63	31.5	5300	3.278	17375	23006	40381	5591
Q-2	63	07 F	1 ( 2 0 0	2 / 5 /	54400	77044	12/573	44.04
Q-3	63	97.5	10.388	3+424	80000	11904	104072	4400
0 1	0) 44	72 0	10005	2 727	46063	69139	114201	5507
	04 64	12.0	12325	20121	40000	00100	114201	1000
0-2	64	270-0	47565	4,150	197403	283280	480683	5475
0-4	64	21000	47202			200200		2112
0-1	65	211.5	36943	4,219	155860	227791	383651	1183
0-2	65		nar ≈					
$\bar{Q}-\bar{3}$	65	234.0	39140	4.723	184850	241665	426515	-21
0-4	65							

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 0			FLIGHT TEST				
		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL AR S	LABOR + BURDEN \$	ENGR MATL
Q-1	66 66	225.0	39004	4.903	191238	281497	472735	
Q-3	66	142.5	23976	5.036	120754	173300	294054	
TO	TAL	1320.0	227022		990977	1402822	2393799	277387

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## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

FLIGHT TEST

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJASSY 0

		МРС	OTHER COST	SUB Total	GEA	TOTAL COST
Q-1	58			•		
Q-2	58					
Q-3	58					
Q=4	58					
0-2	27 50					
Q-3	59				·	
0-4	59					
0-1	60					
Q-2	60					
Q-3	60					
Q-4	60					
Q = 1 Q = 2	61					
0-3	61	5		2200	20	2220
Q-4	61	<u>,</u>		2000	96	2,330
0-1	62	488	34	24678	414	25092
Q-2	62					
Q-3	62	3855	15	79591	1336	80927
Q-4	62					
Q-1	63	5507		101803	1702	103505
0-3	63	4419	-49	183904	2072	196 977
Q-4	63	7717	- <b>- - -</b>	10.004	5015	100011
Q-1	64	5871		175143	3727	178870
Q-2	64					
Q-3	64	19920		555359	11817	567176
Q-4	64					
Q-1 0-2	65 4 E	3539		399022	10646	409668
Q-2	02 65	28		426262	11272	437636
0-4	65	50		720203	77313	DEDICE

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	4-SYSTEM 5-SUBSYSTEM 6-MAJ ASSY		SYSTEM 4 FLIGHT SUBSYSTEM 41 MAJ ASSY 0			TEST		
		MPI	C	OTHER COST	SUB TOTAL	GεΔ	TOTAL Cost	
Q = 1	66 66				472735	14238	486973	
Q-3	66				294054	8856	302910	
TOT	AL	43560	5		2714752	67220	2781972	

## COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT TEST

			DESIGN /ENGR HOURS DOLLARS	PROD Hours Dollars	TEST /QC HOURS DOLLARS	TOTAL HOURS DOLLARS
DESIGN/ENGINEE	RING	•	2271271		7882	2279153
LABOR AT \$	5.601		12734722		31041	12765763
ENGR BURDEN	AT \$	6.677	15183320		35715	15219035
PRODUCTION				11030		11030
LABOR AT \$	3.225			355 <b>7</b> 5		35575
SHOP SUPPORT			273288		28298	301586
LABOR AT \$	4.042		1123021		96006	1219027
PLANN ING					77	77
LABOR AT \$	3.883				299	299
TEST/QC			85457	281912	6730	374099
LABOR AT \$	4.665		387151	1333182	25002	1745335
MFG BURDEN	AT \$	5.899	1917005	1978594	155605	4051204
ENGR MATERIAL			1299684		277387	1577071
MFG MATERIAL				144913		144913
MPC			208712	24321	43566	276599
OTHER COST			1200382	1896		1202278
SUB-TOTAL			34053997	3518481	664621	38237099
GEN & ADMIN			906766	95793	13392	1015951
TOTAL COST			34960763	3614274	678013	39253050

TIME-PHASED COST				
DETAIL - SEE PAGE I	I <b>-</b> 343	II-352	II-356	II-366

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## DESIGN/ENGINEERING

		<b>U</b>		
4-SYSTEM	4		mpdm	2
5-SUB SYSTEM	41	FLIGHT	TFOT	
SUBD OF WORK	DESIGN/ENGINEERIN	G		

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLLARS	LABOR + BURDEN \$
Q <b>-1</b>	58	7.5.	1359	4.574	6216	6184	12400
Q-2	58			<i>,</i>			
Q-3	58	36.0	6104	4.270	26065	23802	4986 <b>7</b>
Q-4	58						
0-1	59	39.0	6549	4.213	27589	22421	50010
Q-2	59						
Q-3	59	34.5	6025	4.188	25231	21935	47166
Q-4	59						
Q-1	60	27.0	4788	4.556	21815	17784	39599
Q-2	60						
Q-3	60	9.0	1479	4.766	7049	5409	12458
Q-4	<b>6</b> 0						
Q-1	61	34.5	5837	4.853	28327	19500	41821
Q-2	61	10 0			<b>a</b>	2/105	71.00/
<u>u</u> -3	61	42.0	1491	4.919	30881	34125	11006
Q-4	61	102 5	17700	5 03/	00507	01 720	171944
	62	102.5	11102	2+054	10000	01124	1/1240
	62	152 0	25500	5 030	129024	1 20 717	257612
0-0	02 40	100.0	20109	9.090	120720	120711	201045
0 - 1	62	186 0	21720	5 184	164440	171628	336068
0-2	63	TOOPO	DIJEU	20104	101110	111020	00000
0-3	63	97.5	16277	5,195	84555	91 958	176513
0-4	63	71.02	IOLII				110010
0-1	64	82.5	14015	5-436	76190	85717	161907
0-2	64			50,50			
0-3	64	285.0	50089	5.246	262782	324091	586873
Q-4	64						
$\tilde{0}-1$	65	2364.0	409832	5,555	2276787	2724247	5001034
Q-2	65						

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING 4-SYSTEM 4 FLIGHT TEST 5-SUBSYSTEM 41 FLIGHT TEST SUBD OF WORK DESIGN/ENGINEERING

		MAN- MON THS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 Q-4	65 65	2977.5	500 26 4	5.708	2855708	3185635	6041343
Q-1 Q-2	66 66	3690.0	639503	5.759	3682749	4665523	8348272
Q-3	66	3135.0	526562	5.572	2933905	3572905	6506810
тот	TAL	13303.5	2271271		12734722	15183320	27918042

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 4 FLIGHT TEST 5-SUBSYSTEM 41 SUBD OF WORK DESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 64 Q-4 64	4.5	816	4.602	3755	5974	9729
Q - 1 65	537.0	93192	3.708	345523	524329	869852
Q-3 65 Q-4 65	258.0	43233	3.668	158582	285658	444240
Q-1 66	391.5	6 <b>79</b> 66	3.599	244622	507271	751893
Q-3 66	405.0	68081	5.443	370539	593 <b>773</b>	964312
TOTAL	1596.0	273288		1123021	1917005	3040026

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# TEST/QC

FLIGHT TEST

4-SYSTEM 4 5-SUBSYSTEM 41 SUBD OF WORK DESIGN/ENGINEERING

	MAN- MON TH S	LABOR HOURS	L'ABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 64	3.0	428	3.652	1563		1563
Q-4 64			•			
Q-1 65	108.0	18819	4.319	81272		81272
Q-2 65						
Q-3 65	109.5	18472	4.412	81502		81502
Q-4 65						
Q-1 66	153.0	26416	4.758	125689		125689
Q-2 66						
Q-3 66	127.5	21322	4.555	97125		97125
TOTAL	501.0	85457		387151		387151

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

FLIGHT TEST

4-SYSTEM 4 5-SUBSYSTEM 41 SUBD OF WORK DESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOUR S	LABOR	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$	ENGR MATL
Q-1	58	7.5	1359	4.574	6216	6184	12400	
Q-2	58							
Q-3	58	36.0	6104	4.270	26065	23802	4986 <b>7</b>	
Q-4	58			ι.				
0-1	59	39.0	6549	4.213	27589	22421	50010	
Q-2	59							
Q-3	59	34.5	6025	4.188	25231	21 935	47166	
Q <b>-4</b>	59							
Q-1	60	27.0	4788	4.556	21815	17784	39599	
Q-2	60							
Q-3	60	9.0	1479	4.766	7049	5409	12458	
Q-4	60							
Q-1	61	34.5	583 <b>7</b>	4.853	28327	19500	47827	
Q-2	61							
Q-3	61	42.0	7497	4.919	36881	34125	71006	
0-4	61							
Q-1	62	103.5	17782	5.034	89507	81739	171246	
ର–2	62							
Q-3	62	153.0	25589	5.038	128926	128717	257643	
Q-4	62							
Q-1	63	186.0	31720	5.184	164440	171628	336068	
Q-2	63							
Q-3	63	97.5	16277	5.195	84555	91 958	176513	
Q-4	63							
Q-1	64	82.5	14015	5.436	76190	85717	161907	
Q-2	64							
Q-3	64	292.5	51333	5.223	268100	330065	598165	6105
Q-4	64							
Q <b>-1</b>	65	3009.0	521843	5.181	2703582	3248576	5952158	180299
Q-2	65							
Q-3	65	3345.0	561969	5.509	3095792	3471293	6567085	322489
0 - 4	65							

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TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM45-SUBSYSTEM41SUBDOFWORKDESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DCLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENG MATI
Q-1 66	4234.5	733885	5.523	4053060	5172794	9225854	31803
Q = 2 66 Q = 3 66	3667.5	<del>6</del> 15965	5.522	3401569	4166678	7568247	47275
TOTAL	15400.5	2630016		14244894	17100 325	31345219	129968

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 FLIGHT TEST 5-SUBSYSTEM 41 SUBD OF WORK DESIGN/ENGINEERING

		MFG MATL	TOTAL Material	MPC	WIND	OT HER COST	TOTAL O/C \$	SUB TOTAL
Q-1	58							12400
Q-2	58							
Q-3	58							49867
Q-4	58							
0-1	59							50010
Q-2	59				i.			
Q-3	59							47166
Q-4	59							20550
	- 60 - 7 0							38222
0-2	<b>6</b> 0							12450
0-4	60 60							12400
$\tilde{0}-1$	61							47877
Q-2	61							,
$\tilde{Q}-3$	61							71006
Q-4	61							
Q-1	62							171246
Q-2	62							
Q-3	62							257643
Q-4	62							
Q-1	63							336068
Q-2	63							
Q-3	63				•			176513
Q-4	63							1/1007
0-1	04							101301
0-2	04 67		6105	2221		54 34 4	54246	660937
0-4	64		0105	<i>L. L.L.</i> L		24240	74740	10000
0-1	65		180299	53927		167388	167388	6353772
Q-2	65			- , , , , , , , , , , , , , , , , , , ,				
$\overline{0}$ - 3	65		322489	57530		289587	289587	7236691
Q-4	65							

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM4FLIGHT TEST5-SUBSYSTEM41SUBD OF WORKDESIGN/ENGINEERING

	MFG Matl	TOTAL Material	MPC	WIND TUNNEL	OT HER COST	TOTAL 0/c \$	SUI TOTAI
Q-1 66		318035	65960		475620	475620	1008546
Q-2 66 Q-3 66		472756	29074		213441	213441	828351
TOTAL		1299684	208712		1200 382	1200382	3405399

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTE	M	4	TT.TCUM	macm
5-SUBSY	STEM	41	TUTGUT	TTOT.
SUBD OF	WORK	DESIGN/ENGINEERI	NG	

					TOTAL
		G	3	А	COST
Q-1	58				12400
Q-2	58				
Q-3	58				49867
Q-4	58				
Q-1	59				50010
Q-2	59				
Q-3	59				47166
Q-4	59				
Q-1	60		75	5	40354
Q-2	60				
Q-3	60		23	7	12695
Q-4	60			_	
Q - 1	61		84	-7	48674
Q-2	61		~ ~	-	
Q = 3	61	1	32	C	72326
0-4	61	~	~ ~		
	62	2	81	4	174120
Q-2	62	,	~ ~		
6-3	62	4	32	4	261967
Q-4	62	-		~	212101
	63	2	01	8	341686
	63	~	05		1704 (4
0 4	()	2	90	1	179404
0-1	03	2	1. 1.	c	145253
	04 4 4	2	44	2	102002
	04 4 /	1.4	<u>م</u> ۲	1	474000
ロース	04 47	14	vc	1	014090
√-4 ∩_1	65	160	51	0	6522200
0-2	45	109	1	0	0722240
w-2 0-2	00 45	107	07		7420765
	6J 65	7.20	01	<b>~</b> ¶	1427103
₩ <b></b> .4	U J				

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 SUBD OF WORK DESIGN/ENGINEERING

	GεA	TOTAL COST
Q-1 66	258268	10343737
Q-3 66	249474	8532992
TOTAL	<b>9067</b> 66	34960763

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### PRODUCTION

4-SYSTEM4FLIGHT TEST5-SUBSYSTEM41SUBD OF WORK PRODUCTION

	MAN-	LABOR	LABOR	LABOR	<b>BUR DEN</b>	LABOR +
	MONTHS	HOURS	RATE	DOLLARS	DOLL ARS	BURDEN \$
Q-3 63					52259	52259
Q-4 63						
Q-1 64					40112	40112
Q-2 64						
Q-3 64	19.5	3486	3.201	11157	256350	267507
Q-4 64						
Q-1 65	31.5	5515	3.224	17781	344305	362086
Q-2 65						
Q-3 65	12.0	2029	3.271	6637	398445	405082
Q-4 65						
Q-1 66					542664	542664
Q-2 66						
Q-3 66					344459	344459
TOTAL	63.0	11030		35575	1978594	2014169

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# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

FLIGHT TEST

# TEST/QC

4-SYSTEM 4 5-SUBSYSTEM 41 SUBD OF WORK PRODUCTION

		MAN- MON THS	LABOR HOUR S	LABOR	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	63	60.0	10129	3.481	35264		35264
Q-4	63						
Q-1	64	39.0	6630	3.903	25878		258 <b>78</b>
Q-2	64						
Q-3	64	214.5	37766	4.275	161462		161462
Q-4	64						
Q-1	65	276.0	47962	4.557	218586		218586
Q-2	65						
Q-3	65	355.5	59778	4.869	291081		291081
Q-4	65						
Q-1	66	417.0	72398	4.998	361871		361871
Q-2	66						
Q-3	66	281.5	47249	5.059	239040		239040
тот	AL	1643.5	281912		1333182		1333182

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# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

FLIGHT TEST

4-SYSTEM 4 5-SUBSYSTEM 41 SUBD OF WORK PRODUCTION

	MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	MFG Matl
Q-3 63	60.0	10129	3.481	35264	52259	87523	
Q-4 63							
Q-1 64	39.0	6630	3.903	25878	40112	65990	
Q-2 64				•			
Q-3 64	234.0	41252	4.185	172619	256350	428969	-45
Q-4 64							, . <u>.</u>
Q-1 65	307.5	53477	4.420	236367	344305	580672	26302
Q-2 65							
Q-3 65	367.5	61807	4.817	297718	398445	696163	81546
Q-4 65							6
Q-1 65	417.0	72398	4.998	361871	542664	904535	-2458
Q-2 65							Apres 1 12" And
Q-3 66	281.5	47249	5.059	239040	344459	583499	39568
TOTAL	1706.5	292942		1368757	1978 594	3347351	144913

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

FLIGHT TEST

4-SYSTEM 4 5-SUBSYSTEM 41 SUBD OF WORK PRODUCTION

		MPC	OTHER COST	SUB TOTAL	G&A	TOTAL COST
Q-3	63			87523	1463	88986
Q-4	63					
Q-1	64			65 990	1404	67394
Q-2	64					
Q-3	64	-16		428908	9127	438035
Q-4	64					
0-1	65	7867	39	614880	16405	631285
Q-2	65					
Q-3	65	14547	826	793082	21373	814455
Q-4	65					
Q-1	66	-510	1023	902590	27183	929773
Q-2	66					
Q-3	66	2433	8	625508	18838	644346
тот	TAL	24321	1896	3518481	95793	3614274

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## DESIGN/ENGINEERING 4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT TEST SUBD OF WORK TEST/QC

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58						
Q-2	58						
Q-3	58						
Q-4	58						
Q-1	59						
Q-2	59						χ.
Q-3	59						
Q-4	59						
Q-1	60						
Q-2	60						
Q-3	60						
Q-4	60						
Q-1	61						
Q-2	61						
Q-3	61	1.5	302	3.460	1045	1182	2227
Q-4	61						
Q-1	62	4.5	784	4.061	3184	3623	6807
Q-2	62						
Q-3	62	4.5	729	4.255	3102	3748	6850
Q-4	62	_					
Q-1	63	7.5	1234	4.219	5206	6506	11712
Q-2	63						
Q-3	63		52	6.904	359	92	451
Q-4	63						
	64	3.0	545	3.776	2073	2870	4943
Q-2	64	2 2					
Q-3	04	3.0	582	4. (84	2784	3718	6502
Q−4 0_1	04 4 =	1 5	337	7 005	3010		~
0 7	0) 4 E	1.2	331	3.985	1343	1939	3282
~~ <u> </u>	60						

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## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESIGN/ENGINEERIN	G	
4-SYSTEM	4		
5-SUBSYSTEM	41	FLIGHT	TEST
SUBD OF WORK	TEST/QC		

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	65	3.0	426	4.345	1851	1943	3794
Q-4 Q-1	65 66	15.0	2545	3.495	8896	8 897	17793
Q-2 Q-3	66 66	1.5	342	3.503	1198	1197	2395
тот	T AL	45.0	7882		31041	35715	66756

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# SHOP SUPPORT 4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT TEST SUBD OF WORK TEST/QC

		MAN- MONTHS	LABOR HOURS	LABOR RATE	L ABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-1	59						
Q-2	59						
Q-3	59						
Q-4	59						
Q-1	60			•			
ଦ-2	60						
Q-3	60						
Q-4	60						
Q-1	61						
Q-2	61						
Q-3	61		2	2.500	5	7	12
Q-4	61						
Q <b>-1</b>	62	9.0	1596	3.073	4905	6048	10953
Q-2	62						
Q-3	62	16.5	2797	2.883	8065	11573	19638
Q-4	62						
Q-1	63	18.0	3120	2.952	9209	16490	25699
Q-2	63						
Q-3	63	24.0	3969	3.353	13307	25623	38930
Q-4	63	•					
Q-1	64	30.0	5026	3.513	17655	25156	42811
Q-2	64	~ ~ ~					
Q-3	64	31.5	5500	3.888	21382	36232	57614
Q-4	64						
Q-1	65	16.5	2912	3.205	9333	13936	23269
<b>Q-2</b>	65	• • •					
1-3	05	19.5	3160	3.612	11415	19450	308 <b>65</b>
0-4	55 ()	•					
Q-1	00	1.5	210	3.343	702	1086	1788
Q = 2	65						

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 4 FLIGHT TEST 5-SUBSYSTEM 41 FLIGHT TEST SUBD OF WORK TEST/QC

	MAN- MONTHS	LABOR HOUR S	LABOR	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-3 66		6	4.667	28	4	32
TOTAL	166.5	28298		96006	155605	251611

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# PLANNING

4-SYSTEM 4 5-SUBSYSTEM 41 SUBD OF WORK TEST/QC

FLIGHT TEST

	MAN- MONTHS	LABOR HOUR S	LABOR	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1 63		3	2.333	7	10	17
Q-2 63 Q-3 63		74	3.946	292	-10	282
TOTAL		77		299		299

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## TEST/QC 4-SYSTEM 4 5-SUBSYSTEM 41 SUBD OF WORK TEST/QC

FLIGHT TEST

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-1	62		70	2.929	205		205
Q-2	62						
Q-3	62		101	3.119	315		315
Q-4	62						
Q-1	63	6.0	943	3.131	2953		2953
Q-2	63						
Q-3	63	13.5	2164	3.413	7386		7386
Q-4	63						
Q-1	64		120	3.808	457		457
Q-2	64						
Q-3	64	12.0	2210	4.108	9079		9079
Q-4	64						
Q-1	65		43	4.744	204		204
Q-2	65						
Q-3	65	6.0	1033	4.103	4238		4238
Q-4	65						
Q-1	66		45	3.711	167		167
Q-2	66						
Q-3	<b>6</b> 3		1	1.999	-2		-2
тот	AL	37.5	6730		25002		25002

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

FLIGHT TEST

4-SYSTEM 4 5-SUBSYSTEM 41 SUBD OF WORK TEST/QC

		MAN- MONTHS	LABOR	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$	ENGR MATL
Q <b>-</b> 1	58							
Q-2	58							
Q-3	58							
Q-4	58							
Q-1	59							
Q-2	59		· •					
Q-3	59							
Q-4	59							
Q-1	60							
0-2	60							
Q-3	60							
Q-4	60							
Q-1	61							
Q-2	61							
Q-3	61	1.5	304	3.454	1050	1189	2239	56
()-4	61			~ ~ ~ ~ ~				
	62	13.5	2450	3.385	8294	9671	17965	6191
0-2	62	21.0	7477	3 177	11493	10001	07.000	10010
0-4	0Z 62	21•V	3021	001 eC	11482	15321	20803	48918
0 - 1	52	21 5	5300	2 279	17275	22004	40201	55015
0-2	63	- <b>1</b> + <b>1</b>	0.00	20220	11010	. 20000	40301	22712
0-3	63	37.5	6259	3-410	21344	25705	47049	44862
0-4	63		and and all of					1.002
0-1	64	33.0	5695	3.544	20185	28026	48211	55071
0-2	64			<b>.</b>				
Q-3	64	46.5	8292	4.009	33245	39950	73195	54756
Q-4	64							
Q-1	65	18.0	3292	3.305	10880	15875	26755	11832
Q-2	65						· · ·	
Q-3	65	28.5	4619	3.790	17504	21393	38897	-214
Q-4	65							

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## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 SUBD OF WORK TEST/QC

FLIGHT TEST

	MAN- MONTHS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
Q = 1 66	16.5	2800	3.488	9765	9983	19748	
Q-3 66	1.5	349	3.507	1224	1201	2425	
TOTAL	249.0	42987		152348	191320	343668	277387

	4-SYSTEM 5-SUBSYSTEM SUBD OF WORK		4 41 TEST/QC	FLIGH	FLIGHT TEST	
		мрс	OTHER COST	SUB Total	G & A	TOTAL Cost
0-1	58		·			
0-2	58					
Q-3	58					
Q-4	58					
Q-1	59					
Q-2	59					
Q-3	59					
Q-4	59					
Q-1	60		-			
Q-2	60					
Q-3	60					
Q-4	60					
Q = 1	61					
Q-2	61	F		222		يعتم المعرف المع
0-6	01 61	כ		2300	38	2338
0-1	62	495	34	24679	414	25002
0-2	62	-00		24010	414	20092
0-3	62	3855	15	79591	1336	80927
Q-4	62		£.27		1000	00721
0-1	63	5507		101803	1702	103505
Q-2	63					
Q-3	63	4419	-49	96281	1610	97891
Q-4	63					
Q-1	64	5871		109153	2323	111476
Q-2	64					
Q-3	64	19920		147871	3146	151017
Q-4	64			-		
Q-1	65	3539		42126	1124	43250
Q-2	65					، محمد بين
Q-3	65	-38		38645	1031	39676
Q-4	65					

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

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# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 SUBD OF WORK TEST/QC

FLIGHT TEST

	MPC	OTHER COST	SUB Total	GεA	TOTAL COST
Q-1 66 Q-2 66			19748	595	20343
Q-3 66			2425	73	2498
TOTAL	43566		664621	13392	678013

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESIGN/ENGINEERING
4-SYS TEM	4
5-SUB SYSTEM	41
FL IGHT TES	ST

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1	58	7.5	1359	4.574	6216	6184	12400
Q-2	58						
Q-3	58	36.0	6104	4.270	26065	23802	49867
0-4	58						
Q-1	59	39.0	6549	4.213	27589	22421	50010
Q-2	59						
Q-3	59	34.5	6025	4.188	25231	21935	47166
Q-4	59						
Q-1	60 60	27.0	4788	4.556	21815	17784	39599
Q-2 0-2	40	0.0	1 . 70			_	
Q-3	60	9.0	1479	4. 156	1045	5409	12458
0-4	00		5 A 4 5				
- 1 0 0	01	34+5	5837	4.853	28327	19500	47827
Q-2	01						
Q-3	61	43.5	7799	4.863	37926	35307	73233
Q-4	61						
Q-1	62	109.5	18566	4.993	92691	85362	178053
Q-2	62						
Q-3	62	156.0	26318	5.017	132028	132465	264493
Q-4	62						
Q-1	63	193.5	32954	5.148	169646	178134	347780
Q-2	63						
Q-3	63	97.5	16329	5.200	84914	92050	176964
Q-4	63						
Q-1	64	85.5	14564	5.374	78263	88587	166850
Q-2	64						
0-3	64	288.0	50671	5.241	265566	327809	593375
Q-4	64						······································
Q-1	65	2367.0	410169	5,554	2278130	2726186	5004316
0-2	65						

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESIGN/ENGINEERING
4-SYSTEM	4
5-SUB SYSTEM	41
FL IGHT TE	ST

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 0-4	65 65	2980.5	500690	5.707	2857559	3187578	6045137
Q-1 Q-2	66 66	3703.5	642048	5.750	3691645	4674420	8366065
Q-3	66	3136.5	526904	5.570	2935103	3574102	6509205
тот	AL	13348.5	2279153		12765763	15219035	2 79 84 79 8

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## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PRODUCTION 4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT TEST

		MAN- MON THS	LABOR HOURS	LABOR PATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-3	63					52 259	52259
Q-4	63						
Q <b>-1</b>	64					40112	40112
Q-2	64						
Q-3	64	19.5	3486	3.201	11157	256350	267507
Q-4	64						
Q-1	65	31.5	5515	3.224	17781	344305	362086
Q-2	65						
Q-3	65	12.0	2029	3.271	6637	398445	405082
Q-4	65						
Q-1	66					542664	542664
Q-2	66						
Q-3	66					344459	344459
TOT	TAL	63.0	11030		35575	1978594	2014169

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP	SUPPORT
4	

4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT TEST

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-1	59						
Q-2	59						
Q-3	59						
Q-4	59						
Q-1	60						
Q-2	60						
Q-3	60						
Q-4	60						
Q-1	61						
Q-2	61						
Q-3	61		2	2.500	5	7	12
Q-4	61						
Q-1	62	9.0	1596	3.073	4905	6048	10953
Q-2	62						
Q-3	62	16.5	2797	2.883	8065	11 573	19638
Q-4	62						
Q-1	63	18.0	3120	2.952	9209	16490	25699
Q-2	63						
Q-3	63	24.0	3969	3.353	13307	25623	38930
Q-4	63						
Q-1	64	30.0	5026	3.513	17655	25156	42811
Q-2	64						
Q-3	64	36.0	6316	3.980	25137	42206	67343
Q-4	64						
Q-1	65	555.0	96104	3.692	354856	538265	893121
Q-2	65						
Q-3	65	276.0	46393	3.664	169997	305108	475105
Q-4	65						
Q-1	66	393.0	68176	3.598	245324	508357	753681
Q-2	66						

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 4 5-SUB SYSTEM 41 FL IGHT TEST

LABOR + BURDEN \$	BUR DÉN DOLLARS	LABOR DOLLAR S	LABOR RATE	LABOR HOUR S	MAN- MON THS	
964344	593777	370567	5.443	68087	405.0	Q-3 66
3291637	2072610	1219027		301586	1762.5	TOTAL

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## PLANNING

4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT TEST

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	MAN- MON THS	LABOR	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-1 63		3	2.333	7	10	17
Q-2 63						
Q-3 63		74	3.946	292	-10	282
Q-4 63						
Q-1 64						
Q-2 64						
Q-3 64						
Q-4 64						
Q-1 65						
Q-2 65						
Q-3 65						
Q-4 65						
Q-1 66						
Q-2 66						
Q-3 66						
TOTAL		77		299		299
NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT TEST

		MAN- MON TH S	LABOR	LABGR RATE	LABOR	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1	62		70	2.929	205		205
Q-2	62						
Q-3	62		101	3.119	315		315
Q-4	62	( )	0.4.2	2 1 2 1	2052		2052
Q-1	63	0.0	943	3.131	295.3		2953
Q-3	63 63	73.5	12293	3.46.9	42650		42650
Q-4	63						
Q-1	64	39.0	6750	3.901	26335		26335
Q-2	64						
Q-3	64	229.5	40404	4.260	172104		172104
Q-4	64						
Q-1	65	385.5	66824	4.490	300062		300062
Q-2	65						
Q-3	65	472.5	79283	4.753	376821		376821
Q-4	65						
Q-1	66	570.0	98859	4.934	487727		487727
Q-2	66						
Q-3	66	408.0	68572	4.902	336163		336163
тот	AL	2184.0	374099		1745335		1745335

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 FL IGHT TEST

		MAN-	LABOR	LABOR	LABOR	BUR DEN	LABOR +	ENGR
		MUNTHS	HOURS	RATE	DULLARS	DULLARS	BURDEN \$	MAIL
0-1	58	7.5	1359	4.574	6216	6184	12400	
Q-2	58							
Q-3	58	36.0	6104	4.270	26065	23802	49867	
0-4	58							
Q-1	59	39.0	6549	4.213	27589	22421	50010	
Q-2	59							
Q-3	59	34.5	6025	4.188	25231	21 93 5	47166	
Q-4	59							
Q-1	60	27.0	4788	4.556	21815	17784	39599	
Q-2	60							
Q-3	60	9.0	1479	4.766	7049	5409	12458	
Q-4	60							
Q-1	61	34.5	5837	4.853	28327	19500	47827	
Q-2	61							_
Q-3	61	43.5	7801	4.862	37931	35314	73245	56
Q-4	61							
Q-1	62	118.5	20232	4.834	97801	91410	189211	6191
Q-2	62							
Q-3	62	172.5	29216	4.806	140408	144038	284446	48918
Q-4	62							
Q-1	63	217.5	37020	4.911	181815	194634	376449	55915
Q-2	63							
Q-3	63	195.0	32665	4.322	141163	169922	311085	44862
Q-4	63		<b>.</b>					55071
Q-1	64	154.5	26340	4.641	122253	153855	276108	55071
Q-2	64					121215	1100000	(00(1
Q-3	64	573.0	100877	4.698	473964	626365	1100329	00301
Q-4	64				0050000	2/0075/	( FEOEDE	102121
Q-1	65	3339.0	578612	5.100	2920,829	3008 136	0707000	195191
Q-2	65		(00005	E (00	2/1101/	2001121	7202145	200075
Q-3	65	3741.0	628395	5.428	5411014	2041131	1202142	526215
Q-4	65							

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT TEST

·	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
Q = 1 66	4666.5	809083	5.469	4424696	5725441	10150137	318035
Q-3 66	3949.5	663563	5.488	3641833	4512338	8154171	472756
TOTAL	17358.0	2965945		15765999	19270239	35036238	1577071

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### 4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT TEST

		MFG Matl	TOTAL MATERIAL	MPC	WIND TUNNEL	OT HER COST	TOTAL D/C \$	SUE TOTAL
Q-1	58							12400
Q-2	58							×
Q-3	58							49867
Q-4	58							
	59							50010
	59							47166
0-4	59							41100
0-1	60					·		39599
0-2	60							
$\alpha - 3$	60							12458
Q-4	60							
Q-1	61							47827
Q-2	61							
Q-3	61		56	5				73306
Q-4	61							
Q-1	62		6191	488		34	34	195924
Q-2	62							222024
Q-3	62		48918	3855		15	15	331234
Q-4	62		55015	5507				427971
	0.3 4 2		22412	2201				451011
0-2	6.0		44862	4419		-49	-49	360317
Q-5 0-4	63	•	44002				12	500511
0-1	64		55071	5871				337050
0-2	64							
Q-3	64	-45	60816	22125		54346	54346	1237616
Q-4	64							
Q-1	65	26302	218433	65333	ň	167427	167427	7010778
Q-2	65							
Q-3	65	81546	403821	72 03 9		290413	290413	8068418
Q-4	65							

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT TEST

SUB Total	TOTAL O/C \$	OTHER COST	WIND TUNNEL	MPC	TOTAL MATERIAL	MFG Matl	
11007807	476643	476643		65450	315577	-2458	Q-1 66
8911451	213449	213449		31507	512324	39568	Q-3 66
38237099	1202278	1202278		276599	1721984	144913	TOTAL

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

TIME	PHASED	EXPEND.
B-70	AIRCRAF	T STUDY

4-SYSTEM	4
5-SUB SYSTEM	41
FL IGHT TE	ST

		GEA	TOTAL COST
Q-1	58		12400
Q-2	58		
Q-3	58		49867
Q-4	58		
Q-1	59		50010
Q-2	59		
Q-3	59		47166
Q-4	59		
Q-1	60	755	40354
Q-2	60		
Q-3	60	237	12695
Q-4	60		
Q-1	61	847	48674
Q-2	61		
Q-3	61	1358	74664
Q-4	61		
Q-1	62	3288	199212
Q-2	62		
Q-3	62	5660	342894
Q-4	62		
Q-1	63	7320	445191
Q-2	63	(	21/212
Q-3	63	6024	366341
Q-4	63	7170	2// 202
Q-1	64	1112	344222
Q-2	84	21221	12/2050
Q-3	04	20334	1203950
<b>₩</b> <sup></sup> 4 01	04 45	197047	7107925
Q-1 0-2	00 65	101041	1121053
<b>₩</b> <sup>-</sup> 2	09 65	215479	8282806
Q-2	65	217410	0203030
	U J		

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT TEST

	GδA	TOTAL COST
Q = 1 66	286046	11293853
Q-3 66	268385	9179836
TOTAL	1015951	39253050

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#### WBS 4.0 FLIGHT TEST

#### Flight Operations

WBS CODE: 4.41.1

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This section presents an operations engineering summary covering the many facets of Planning, Ground Operations, and Flight Operations that existed during the XB-70 Flight Test Development Program.

#### Planning:

Operations Engineering planning for the support of the two air vehicle program consisted of two Flight Mission Planners in addition to three teams of engineers. Two of these teams were used in direct support of the air vehicles, the third in direct support of the ground support equipment (FTGSE) utilized by the air vehicles. Each air vehicle team consisted of a team leader and five air vehicle system-oriented engineers. The third team consisted of a team leader and four engineers, two assigned responsibility for electrical FTGSE and two assigned responsibility for mechanical FTGSE.

Detailed assignments of the members of each air vehicle team were as follows:

<u>Team Leader</u>: Assigned overall responsibility for the technical direction of all flight and ground test activities, control of all flight and ground operations, coordination and scheduling of support activities for flight and ground tests, and the technical support of systems configurations.

System Engineer No. 1 - Operational support of Landing Gear, Hydraulics, and Accessory Drive Systems.

System Engineer No. 2 - Operational support of Propulsion and Fuel Systems.

System Engineer No. 3 - Operational support of Environmental Control, Escape, and Personnel Accommodations Systems.

System Engineer No. 4 - Operational support of Flight Controls, Flight Augmentation Control, and Air Induction Control Systems.

System Engineer No. 5 - Operational support of Electrical Power Distribution, Communication, Navigation, Cockpit Display, and miscellaneous electronics systems.

Flight Mission Planner - Responsible to plan each mission in its entirety, including ground track, flight profile, fuel consumption, test points to be obtained, and support required, i.e. safety chase air vehicles, communication frequency assignments, restricted area clearance, etc.



Operational Procedures formulated as operational control documents and utilized throughout the flight test program were as follows:

Flight Test Procedure Letters - Procedure Letters were originated to define standard methods of operation pertaining to the XB-70 Project.

Operational Test Plans - Prior-to-flight ground operations for this air vehicle were extensive due to its size and the complexity of its systems. An integrated schedule was developed, and detailed Operational Test Plans (OTP's) were compiled in order to assure timely completion of these operations in a safe manner. In order that these OTP's be uniform and complete, a Flight Test Procedure Letter was generated to define format and content.

Flight Test Technical Guides - Technical Guides, applicable to specific air vehicles or equipment, were prepared as tools to enable operations and maintenance groups to accomplish repetitive shop testing or post- and preflight activities safely, sequentially, and completely. Material contained in these Technical Guides was based on Engineering Process Specifications, released drawings, Engineering Orders, etc., but was presented in a manner that enabled the user to economically and expeditiously accomplish his assigned task. A Flight Test Procedure Letter was prepared to define format and content of the Technical Guide.

Planning for ground operations included the compilation of all requirements from Process Specifications, Test Authorizations, Engineering Orders, etc., and the integration of these requirements into a smooth flowing plan scheduled to preclude delays or lost motion of any kind. The OTP's were prepared to provide for accumulation of parts, test equipment, ground support equipment, personnel, and information at the proper time to integrate the activity. The systems operations engineers were required to coordinate and conduct their assigned activities as prescribed in the OTP's which they had prepared with attention to safe, sequential, and complete accomplishment of their tasks. Planning for flexibility of operations was carried out to provide for delays due to malfunctions. Due to the size of the air vehicle, the noise level in the surrounding area, and the number of personnel involved in the operations, a central communications station was designed, constructed, and employed during the ground test program.

This ground communication station was housed in a 29-passenger bus to give it the mobility required for the program. It contained two-way radio communications, a five-channel ground inter-communication system capable of being tied in to the air vehicle inter-communication system as well as to that used by ground support equipment personnel. Completely selfcontained electrical power generation was provided so that it could operate at remote sites and could maintain radio communications while in motion. It provided seating for driver and ten people and equipment maintenance technician. This Mobile Communication Station proved invaluable



during the course of ground and flight operations.

Planning for flight operations revolved around the Operations team. Specifications were drawn for a Flight Control Communications room, designed to provide for the entire team to be seated at a telemetry and radar console where specific air vehicle flight safety parameters would be displayed. These provided intelligence concerning air vehicle position (ground track and altitude), velocity, attitude, surface controls position, structural stresses, fluid systems pressures, and specific systems health information.

In addition, seats were provided for specific support functions such as Safety Chase Air Vehicle Controller, FAA airspace coordinator, range communications controller, Flight Mission Planner and a member of management to act as final authority for important decisions. In support of this group a second room was specified to display, on analog recorders, the same telemetered intelligence to a group of design engineers. This provided for expert analysis of data, on-the-spot, to support the operations team in decisions pertaining to in-flight malfunctions.

The Flight Communications Control room was connected by a microwave network to the Edwards Flight Test Range (EFTR) communications and tracking stations. The EFTR provided a test mission environment in a highly instrumented flight test corridor extending from Wendover, Utah, to Edwards AFB, California. By properly planning the ground track and flight profile, this communication and telemetry network, together with its tracking radar stations, could be used to maintain constant surveillance of the air vehicle.

Each Flight Test Operations Engineer was required to study all information available pertaining to his systems in order to gain the detailed knowledge of their design and operating characteristics. Each engineer attended such briefings as were available; contacted design personnel cognizant of his systems; studied drawings, technical manuals and reports; and engaged in the operation of simulators and other system testing available. Each was periodically required to brief the entire group on the status of his systems and their operational characteristics. This provided a degree of cross-training as well as the systems interface information necessary to the welding of the individual engineers into an operating team.

As manufacturing of the air vehicle approached completion, the team was employed in the final operations checkout and assisted in problem analysis and resolution. This application of their knowledge enhanced their training and led directly into the Ground Test Operations required to proof test each system prior to flight. Additional training was acquired by reviewing Safety Analyses and conducting additional analysis wherever indicated.



Since the method to be used for flight monitoring was somewhat advanced in respect to previous methods, a simulated flight mission was conducted from the Flight Communications Control room, using a TB-58 as the flight vehicle. This exercise served a two-fold purpose: It gave the team an opportunity to acquire first-hand experience in the use of the Control Room while conducting a mission with pre-planned XB-70 malfunction events, and at the same time exercised the functional operation of the Control Room and its equipment.

Ground Operations:

Complete ground testing of the air vehicle and all of its subsystems was accomplished upon completion of its manufacture and prior to its flight operations. The Operational Test Plans, as noted above, were integrated by schedule and accomplished to ensure satisfactory operation of each system, individually, and interfaced with all associated systems. They were conducted by the Flight Test Operations Teams, assigned to cover two twelve-hour shifts in order to expedite the operation. Malfunctions and other problems were repaired and otherwise resolved on-the-spot by the team whenever possible. When on-the-spot resolution was not possible, coordination with the cognizant group of Engineering or Procurement personnel was initiated and follow-up maintained until the solution was obtained. This method of operation proved very satisfactory from a point of view of expeditious completion of the ground activities in that it allowed for continuous surveillance of air vehicle activity by knowledgeable systems operation personnel with excellent shift-to-shift coordination. Culmination of this occurred with the taxi tests where all systems were functionally operated together to assure proper operation for flight.

Special tests were conducted periodically throughout the program in order to determine specific characteristics of the air vehicle. For example, after four flights the No. 1 Air Vehicle was returned to Palmdale from Edwards AFB and set up in special fixtures to proof-load test the control surfaces and the inlet duct. These tests were conducted by the Structures Test Lab personnel under the surveillance of Flight Test personnel. The fuel system calibration was another example of such a special test. It was conducted by Flight Test Operations personnel in the Edwards AFB Weight and Balance Facility in accordance with a pre-planned OTP.

A set of post- and preflight requirements was compiled and formalized on prepared forms by the Flight Test Maintenance group with the assistance of the Operations Engineers. These were tailored to integrate air vehicle system checkout and servicing into coordinated operation in order to safely and expeditiously prepare the air vehicle for flight, using the maintenance and operations teams in coordinated effort. These forms were updated periodically during the course of the program as experience was gained and system changes were made.

During ground operations, FTGSE was treated the same as any air vehicle system. Assigned engineers and maintenance personnel maintained constant

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surveillance of air vehicle system changes and up-dated the FTGSE to match all changes made. A program for improving operation and use was constantly progressing throughout the air vehicle test program. The Instrumentation Subsystem (WBS: 1.11), in addition to recording flight parameters for the test program, was used throughout the period to gather information in support of malfunction analysis. It proved an invaluable assistant to Operations and Design engineers in analyzing causes of malfunctions and proving operational suitability of corrections made.

#### Flight Operations:

The Flight Mission Planners custom-tailored the air vehicle flight mission parameters to meet test requirements. They developed the mission profile and ground track to fit the capabilities of the EFTR instrumentation and sequenced the test points to fit the pattern. It was their task to assure proper fuel management, air vehicle weight and balance, mission support activities, and be constantly acquainted with air vehicle system status. Once the mission plan was finalized, they participated in the mission briefing for the pilots and furnished detailed information on conduct of the mission.

During flight, the mission planner was required to monitor all elements of air vehicle situation in order to advise on energy management for return of the air vehicle to Edward AFB. He provided alternate test points that could be accomplished in the event a malfunction or environmental change required a change of mission.

The mission support to flight operations was as follows:

EFTR Facilities: Communications, telemetry, and radar tracking facilities are provided in real-time by the EFTR. For the XB-70 program, two channels of UHF, 36 parameters of telemetered information, and a constant-surveillance radar-tracked ground plot were provided.

Telemetry and radar ground track equipment operation were considered very good. These displays were consistently available and generally accurate. They were occasionally disrupted while flight was in progress, principally at extremes of acquisition capability when the air vehicle was in a turn.

Radio communications were only fair if considered on an overall basis. Frequent loss of communications occurred, principally as a result of low power output equipment at the uprange stations. Adding to this difficulty were the many connections and patch panels in the system associated with the ground station at the Flight Communications Control Room. Backup communications capabilities were provided in three ways. First, a phone patch system was installed to enable the Flight Controller to use transceivers installed at the NAA facility. This system was used with success several times. Second, a relay message system was employed, using a telephone conference call system to the FAA facilities at Salt Lake City, Denver, and Oakland. This



necessarily limited the degree of communication, but proved helpful several times in determining air vehicle status. Third, the NASA High Range Communication System was "piped" into the Control Room. This is associated with, but uses different equipment than, the EFTR. This method was used several times with success. Without these alternate methods of communications, many test missions would have had to be aborted.

<u>Chase Air Vehicles</u>: Safety chase air vehicles were mandatory for every mission. They provided the pilot with "external eyes" from which to view his vehicle.

General practice was to provide a constant surveillance during the flight regime where chase capability matched that of the XB-70. In flight regimes where no chase was available to match the XB-70 capability, a proximity chase was provided to be immediately available should the XB-70 experience a serious problem. The TB-58 was the most versatile of all chase aircraft provided. It had the greatest velocity-range capability and proved invaluable to the program.

Emergency Search and Rescue Procedures: Emergency search and rescue procedures were prepared by Flight Test Operations Engineering, coordinated with Industrial Security (NR) and with the Air Force, and presented in a Flight Test Procedure Letter.

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 01 FLIGHT OPERATIONS

	·		DESIGN ZENGR	PROD	T OT AL
			HOURS	HOURS	HOURS
			DOLLARS	DOLLARS	DOLLARS
DESIGN/ENGINEE	RING		1958503	1	1958503
LABOR AT \$	5.774		11308624		11308624
ENGR BURDEN	AT \$	6.892	13498804		13498804
SHOP SUPPORT			81949		81949
LABOR AT \$	3.713		304273		304273
TEST/QC			15083	107592	122675
LABOR AT \$	4.788		63784	523629	587413
MFG BURDEN	AT \$	5.979	465052	758409	1223461
ENGR MATERIAL			430206		430206
MFG MATERIAL				144913	144913
MPC			75447	24321	99768
OTHER COST			1189794	1896	1191690
SUB-TOTAL			27335984	1453168	28789152
GEN & ADMIN			763700	41560	805260
TOTAL COST			28099684	1494728	29594412

TIME-PHASED COST DETAIL - SEE PAGE II-386

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## DESIGN/ENGINEERING

4-SYSTEM45-SUBSYSTEM416-MAJASSY01SUBDOF WORKDESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN É DOLL ARS	LABOR + BURDEN \$
Q-1	53	7.5	1359	4.574	6216	6184	12400
Q-2	53						
Q-3	58	36.0	5974	4.270	25507	23296	48803
Q-4	58						
Q-1	59	36.0	6234	4.213	26262	21344	47606
Q-2	59						
Q-3	59	33.0	5749	4.185	24062	20929	44991
Q-4	59						
Q-1	60	22.5	3871	4.555	17634	14363	31997
Q-2	60						
Q-3	60	9.0	1439	4.769	6863	5260	12123
ଦ–4	60						
Q-1	61	34.5	5813	4.680	27204	19420	46624
0-2	61						
Q-3	61	33.0	6074	4.903	29780	27638	57418
Q-4	61						
Q-1	62	72.0	12170	5.007	60941	55962	116903
Q-2	62						
Q-3	62	118.5	19969	4.998	99807	100253	200060
Q-4	62	1/0 5	04400			****	
	63	142.5	24422	5.153	125084	132084	258168
Q-2	03	11 5	1000/	£ 33.5	5443.5		110/00
0-3	03	01.0	10394	2.233	24410	27798	113698
0 - 4	03	11 5	7017	5 (07	11202	10507	~~~~~
Q-1 0 0	04	40.0	1911	<b>3.6</b> (21	44393	48527	92920
Q-2	04 47	244 0	62260	5 300	224.050	370433	E 03 30 3
0-1	04	240.0	43240	<b>D</b> •200	224850	218433	203283
Q=4 0.1	04	2060 E	350Z71	5 571	1000010	2202244	1202251
ч <b>—</b> Т	02	2007+2	323011	2.240	エンシンシエク	2072044	4376234

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## DESIGN/ENGINEERING 4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT OPERATIONS 6-MAJ ASSY 01 SUBD OF WORK DESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-2 65						
Q-3 65 Q-4 65	2629.5	441758	5.838	2578772	2862277	5441049
Q-1 66 Q-2 66	3297.0	571606	5.924	3386330	4276741	7663071
Q-3 66	2571.0	431843	5.950	2569599	3154461	5724060
TUTAL	11465.5	1958503		11303624	13498804	24807428

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP	SUPPORT	
4-SYSTEM	4		
5-SUB SYSTEM	41	FLIGHT	OPERATIONS
6-MAJ ASSY	01		
SUBD OF WORK	DESIC	SN/ENGINEERING	

	MAN- MCNTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABUR + Burden \$
Q-3 64 Q-4 64	3.0	582	4.881	2841	4362	7203
Q-1 65 Q-2 65	249.0	43188	3.728	161018	226929	387947
Q-3 65 Q-4 65	75.0	12587	3.457	43518	70702	114220
Q-1 66 Q-2 66	28.5	502 <b>7</b>	3.914	19575	40117	59792
Q-3 66	123.0	20565	3.755	77221	122942	200163
TOTAL	478.5	81949		304273	465052	769325

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## TEST/QC 4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT OPERATIONS 6-MAJ ASSY 01 SUBD OF WORK DESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOURS	LABOR	LABOR Dollars	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 64	1.5	322	3.798	1223		1223
Q-4 64						
Q-1 65	34.5	6046	4.236	25613		25613
Q-2 65						
Q-3 65	19.5	3234	4.049	13096		13096
0-4 65						
Q-1 66	16.5	2785	4.419	12307		12307
0-2 66						
Q-3 66	16.5	2696	4.282	11545		11545
TOTAL	88.5	15083		63784		63784

## **APRIL 1972**

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM45-SUBSYSTEM416-MAJ ASSY01SUBD OF WORKDESIGN/ENGINEERING

		MAN-	LABOR	LABOR	LABOR	BURDEN	LABOR +	ENGR
		MUNIHS	HUURS	RATE	DOLLARS	DOLLARS	BURDEN \$	MATL
0-1	58	7.5	1359	4.574	6216	6184	12400	
Q-2	58						ι.	
0-3	58	36.0	5974	4.270	25507	23296	48803	
Q-4	58							
Q-1	59	36.0	6234	4.213	26262	21344	47606	
0-2	59							
Q-3	59	33.0	5749	4.185	24062	20929	44991	
Q-4	59							,
Q-1	60	22.5	3871	4.555	17634	14363	31997	Ę
Q-2	60							
Q-3	60	9.0	1439	4.769	6863	5260	12123	
Q-4	60							
Q-1	61	34.5	5813	4.680	27204	19420	46624	
$\nabla = Z$	61	22.0	4074	(				
Q-3	01	25.0	6074	4.903	29780	27638	57418	
Q=4 Ω_1	42	72 0	10170	F 007	100/1	FE 64 3	11/000	
Q=1 0-2	62	12.0	12170	5 • 00 T	00941	22202	116903	
0-3	62	118 5	10060	6 00 8	00007	100 252	200010	
<u>∧</u> -4	62	110.0	19909	<b>H + 330</b>	97001	100255	200060	
$\tilde{0}-1$	63	142.5	24422	5 162	126084	122004	250160	
0-2	63			J. 105	120004	1 32 604	2 20100	
$\tilde{0}-3$	63	61.5	10394	5.235	54410	59288	113698	
0-4	63	~~~~~	19271		21110		110000	
0-1	64	46.5	7917	5.607	44393	48527	92920	
0-2	64					• • • • • •	10100	
0 - 3	64	250.5	44144	5.186	228914	282795	511709	5924
Q-4	64					terms of these of 12 and		and the face of
Q <b>-1</b>	65	2353.0	407905	5.360	2186541	2619273	4805814	110670
Q-2	65			_	··· . –			
Q-3	65	2724.0	457579	5.759	2635386	2932979	5568365	<b>9</b> 82 <b>7</b> 3

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM45-SUBSYSTEM41FLIGHT OPERATIONS6-MAJ ASSY01SUBD OF WORKDESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
0-4 65							
Q-1 66	3342.0	579418	5.900	3418312	4316858	7735170	64548
Q-2 66							
Q-3 66	2710.5	455104	5.841	2658365	3277403	5935768	150791
TOTAL	12032.5	2055535		11676681	13963856	25640537	430206

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM45-SUBSYSTEM416-MAJ ASSYC1SUBD OF WORKDESIGN/ENGINEERING

		МРС	WIND TUNNEL	OTHER COST	TOTAL O/C \$	SUB TOTAL	G&A	TOTAL COST
Q-1	<b>5</b> 8					12400		12400
Q-2	58							
Q-3	58					48803		48803
Q-4	58							
Q-1	59					47606		47606
Q-2	59							
Q = 3	59					44991		44991
Q-4 Q-1	<b>59</b> 60					31997	610	32607
Q-2	60							
Q-3	60					12123	231	12354
Q-4	60							
Q-1	61					46624	825	47449
Q-2	61							
Q-3	61					57418	1067	58485
Q-4	61							
Q-1	62					116903	1962	118865
Q-2	62				,	2000/0	-3.750	202/30
8-3	62					200060	3338	203418
0-4	02				•	250140	1.714	243404
0-2	60) 47					200100	4510	202404
0-2	42					112408	1001	115500
Q = 3 $\Omega = 4$	63					113030	1901	TT7533
0-1	64					92920	1977	94897
0-2	64					<i>JE JE</i> V	4.211	24071
0-3	64	2155		54346	54346	574134	12216	586350
0-4	64	E 200		2.3.0	2,2,0	211201		
0-1	65	33101		166018	166018	5115603	136484	5252087
0-2	65					· · ·		
Q-3	65	17531		28858 <b>7</b>	288587	5972756	159353	6132109

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 FLIGHT OPERATIONS 6-MAJ ASSY 01 SUBD OF WORK DESIGN/ENGINEERING

	MPC	WIND TUNNEL	OTHER COST	TOTAL 0/C \$	SUB Total	G&A	TOTAL Cost
0-4 65							
Q-1 66	13387		473588	473588	8286693	249570	8536263
Q-2 66							
Q-3 66	9273		207255	207255	6303087	189830	6492917
TOTAL	75447		1189794	1189794	27335984	763700	28099684

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PRODUCTION

4-SYSTEM45-SUBSYSTEM416-MAJ ASSY01SUBD OF WORKPRODUCTION

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BURDEN DOLLARS	LABOR + BURDEN \$
Q-3	64					13020	13020
Q-4	64						
Q-1	65					131578	131578
Q-2	65						
Q-3	65					170522	170522
Q-4	65						
Q-1	66					271151	271151
Q-2	66						
Q-3	66					172138	172138
TOT	AL					758409	758409

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### PLANNING

4-SYSTEM45-SUBSYSTEM416-MAJ ASSY01SUBD OF WORKPRODUCTION

#### ON-SITE LABOR

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + Burden \$
2-3	64						
2-4	64						
2-1	65						
Q-2	65						
2-3	65						
Q-4	65						
0-1	66						
Q-2	66						
Q-3	66						

TOTAL

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## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC4-SYSTEM45-SUBSYSTEM41FLIGHT OPERATIONS6-MAJ ASSY01SUBD OF WORKPRODUCTION

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 Q-4	64 64	11.5	1979	4.275	8461		8461
Q-1 Q-2	65 65	114.0	19705	4.614	90916		90 <b>916</b>
Q-3 Q-4	65 65	156.0	26126	4.764	124473		124473
Q-1 Q-2	66 66	208.5	36194	4.984	180396		180396
Q-3	66	141.0	2358 <b>8</b>	5.061	119383		119383
тот	AL	631.0	107592		523629		523629

**APRIL 1972** 

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM	4	
5-SUBSYSTEM	41	
6-MAJ ASSY	01	-
SUBD OF WORK	PRODUCTION	

FLIGHT OPERATIONS

l I	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BURDEN DOLLARS	LABOR + BURDEN \$	MFG Matl
Q-3 64 Q-4 64	11.5	1979	4.275	8461	13020	21481	-45
Q-1 65 Q-2 65	114.0	19705	4.614	90916	131578	222494	26302
Q-3 65 Q-4 65	156.0	26126	4.764	124473	170522	294995	81546
Q-1 66 Q-2 66	208.5	36194	4.984	180396	271151	451547	-2458
Q-3 66	141.0	23588	5.061	119383	172138	291521	39568
TOTAL	631.0	107592		523629	758409	1282038	144913

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM	4	
5-SUB SYSTEM	41	FLIGHT OPERATIONS
6-MAJ ASSY	01	
SUBD OF WORK	PRODUCTION	

	мрс	OTHER COST	SUB TOTAL	GδA	TOTAL COST
Q-3 64 Q-4 64	-16		21420	456	21876
Q-1 65 Q-2 65	7867	39	256702	6849	263551
Q-3 65 Q-4 65	14547	826	391914	10670	402584
Q-1 66 Q-2 66	-510	1023	449602	13540	463142
0-3 66	2433	8	333530	10045	343575
TOTAL	24321	1896	1453168	41560	1494728

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING 4-SYSTEM 4 5-SUBSYSTEM 41

6-MAJ ASSY 01 FLIGHT OPERATIONS

		MAN-				BURDEN	LABOR +
		MUN 103	HUUK 3	KAIE	DULLAKS	DULLAKS	DUKDEN Þ
Q-1	58	7.5	1359	4.574	6216	6184	12400
Q-2	58						
Q-3	58	36.0	5974	4.270	25507	23296	48803
Q-4	58						
Q-1	59	36.0	6234	4.213	26262	21344	47606
Q-2	59	20.0					
Q-3	59	33.0	5749	4.185	24062	20929	44991
Q = 4 0 = 1	29	22 5	2071		17676	11.747	21007
Q-1 0-2	60	22.00	5071	4.9995	11034	14305	31331
0-3	60	9.0	1439	4.769	6863	5260	12123
0-4	60		2,33				
Q-1	61	34.5	5813	4.680	27204	19420	46624
Q-2	61						
Q-3	61	33.0	6074	4.903	29780	27638	57418
Q-4	61		•				
Q-1	62	72.0	12170	5.007	60941	55962	116903
Q-2	62		100/0		~~~~	100000	
Q-3	62	118.5	1,4994	4.998	99801	100253	200060
0 - 4	62	142 5	24422	5 163	126.084	132084	258168
0-2	63	172.07	24422	2.107	120004	1 92 00 1	20100
Q-3	63	61.5	10394	5.235	54410	59288	113698
Q-4	63						
Q-1	64	46.5	7917	5.607	44393	48527	92920
Q-2	64						
Q-3	64	246.0	43240	5.200	224850	278 433	503283
Q-4	64				1000010		
Q-1	65	2069.5	358671	5.576	1999910	2392344	4392254

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING 4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 01 FLIGHT OPERATIONS

	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DCLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-2 65						
Q-3 65	2629.5	441758	5.838	2578772	2862277	5441049
Q-4 65						
Q <b>-1 6</b> 6	3297.0	571606	5.924	3386330	4276741	7663071
Q-2 66						
Q-3 66	2571.0	431843	5.950	2569595	3154461	5724060
TOTAL	11465.5	1958503		11308624	13498804	24807428

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### PRODUCTION 4

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJASSY 01 FLIGHT OPERATIONS

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	64					13020	13020
Q-4	64						
Q-1	65					131578	131578
Q-2	65						
Q-3	65					170522	170522
0-4	65						
Q-1	66			•		271151	271151
0-2	66		•				
Q-3	66					172138	172138
тот	T AL					758409	758409

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## SHOP SUPPORT 4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 01 FLIGHT OPERATIONS

		MAN- MON TH S	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL AR S	LABOR + BURDEN \$
Q-3	64	3.0	582	4.881	2841	4362	7203
Q-4	64						
Q <b>-1</b>	65	249.0	43188	3.728	161018	226929	387947
Q-2	65						
Q-3	65	75.0	12587	3.457	43518	70702	114220
Q-4	65						
Q-1	66	28.5	5027	3.914	19675	40117	59792
Q-2	66						
Q-3	66	123.0	20565	3.755	77221	122942	200163
тот	<b>AL</b>	478.5	81949		304273	465052	769325

# **APRIL 1972**

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 01 FL IGHT OPERATIONS

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-3 0-4	64 64	13.5	2301	4.209	9684		9684
Q-1 Q-2	65 65	148.5	25751	4.525	116529		116529
Q-3 Q-4	65 65	175.5	29360	4.686	137569	•	137569
Q = 1 Q = 2	66 66	225.0	38979	4.944	192703		192703
Q-3	66	156.0	26284	4.981	130928		130928
тот	AL	718.5	122675		587413		587413

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 01 FLIGHT OPERATIONS

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		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLLARS	LABOR + BURDEN \$	ENGR MATL
Q-1	58	7.5	1359	4.574	6216	6184	12400	
Q-2	58							
Q-3	58	36.0	5974	4.270	25507	23296	48803	
Q-4	58							
Q-1	59	36.0	6234	4.213	26262	21344	47606	
Q-2	59							
0-3	59	33.0	5749	4.185	24062	-20929	44991	
Q-4	59							
Q-1	60	22.5	3871	4.555	17634	14363	31997	C.
Q-2	60		1 4 2 4			50/0		
Q-3	60	9.0	1439	4.169	686.3	5260	12123	
Q-4	60	~ ~ ~ ~	1 -		2700/	10/00		
	61	34.5	5813	4.680	27204	19420	40624	
Q-2	61		( 0 <b>7</b> /	( 002	20700	77 (70	57410	
Q-3	01 41	0.cc	0074	4.905	29100	21000	27410	
0-1	61	72 0	12170	5 007	60041	55060	116903	
0-2	0Z 40	12.0	12170	0.004	00341	JJ 302	110303	
0	62	118 5	19969	4 008	99807	100 253	200060	
0-4	62	TTO • >	1,20,2	-10 990	//001	100220	200000	
n = 1	63	142.5	24422	5,163	126084	132084	258168	
0-2	63	1,2,0,2	2. 1 1	2.05	120001			
0-3	63	61.5	10394	5.235	54410	59288	113698	
0-4	63		10371					
Q-1	64	46.5	7917	5.607	44393	48527	92920	
Q-2	64							
$\overline{0-3}$	64	262.5	46123	5.147	237375	295815	533190	5924
0-4	64							
0-1	65	2467.0	427610	5.326	2277457	2750851	5028308	110670
Q-2	65							
Q-3	65	2880.0	483705	5.706	2759859	3103501	5863360	98273

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 01 FL IGHT OPERATIONS

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$	ENGR MATL
Q-4 65							
Q-1 66	3550.5	615612	5.846	3598 <b>7</b> 08	4588009	8186717	64548
Q-2 66			5		0	(207000	1 2 2 2 4
Q-3 66	2850.0	478692	5.803	2111148	3449541	6227289	150791
TOTAL	12662.5	2163127		12200310	14722265	26922575	430206

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 01 FLIGHT OPERATIONS

		MFG MATL	TOTAL MATERIAL	MPC	WIND TUNNEL	OT HER COST	TOTAL D/C \$	SUB TOTAL
Q-1	58							12400
Q-2	58							
Q-3	58							48803
Q-4	58							
Q-1 0-2	29 50							47606
N-3	59							44001
0-4	59							77271
$\overline{Q-1}$	60							31997
Q-2	60							Ć
Q-3	60							12123
Q-4	60							
Q-1	61							46624
Q-2	61							<b>57</b> (10
Q-3	61							5/418
0-1	62							116903
$\overline{0-2}$	62							110705
Q-3	62							200060
Q-4	62							
Q-1	63							258168
Q-2	63							
Q-3	63							113698
Q-4	63							
Q = 1	64							92920
Q-2	04 44	-45	5070	2120		54346	54346	505554
0-4	64	-45	2019	2133		0 FC FC	04040	727774
$\tilde{\mathbf{Q}} = 1$	65	26302	136972	40968		166057	166057	5372305
Q-2	65	20202	220972	.0.000			******	~ ~ 1 8 4 7 7 7
Q-3	65	81 546	179819	32 07 8		289413	289413	6364670

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 01 FLIGHT OPERATIONS

.

SUB TOTAL	TOTAL 0/C \$	OTHER COST	WIND TUNNEL	MPC	TOTAL MATERIAL	MFG MATL	
8736295	474611	474611		12877	62090	-2458	Q-4 65 Q-1 66
6636617	207263	207263		11706	190359	39568	Q-2 66 Q-3 66
28789152	1191690	1191690		99768	575119	144913	TOTAL
NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYS	TEM	4	
5-SUB	SYSTE	M 41	
6-MAJ	ASSY	01	
FL	IGHT	OPERAT	IONS

		GεA	TOTAL Cost
Q-1	58		12400
Q-2	58		
Q-3	58		48803
Q-4	58		
Q-1	59		47606
Q-2	59		
Q-3	59		44991
Q-4	59		
Q-1	60	610	3260 <b>7</b>
Q-2	_ <b>6</b> 0		
Q-3	60	231	12354
Q-4	60		
Q-1	61	825	47449
Q-2	61		
Q-3	61	1067	58485
0-4	61		
Q-1	62	1962	118865
Q-2	62		
Q-3	62	3358	203418
Q-4	62		
Q-1	63	4316	262484
Q-2	63		
Q-3	63	1901	115599
Q-4	63		
Q-1	64	1977	94897
Q-2	64		
Q-3	64	12672	608226
Q-4	64		
Q-1	65 ( 5	143333	5515638
Q = 2	65 ( 5	170000	1
Q-3	65	170023	6534693

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 01 FLIGHT OPERATIONS

	GEA	TOTAL Cost
Q-4 65		
Q-1 66	263110	8999405
Q-2 66		
Q-3 66	199875	6836492
TOTAL	805260	29594412



WBS 4.0 FLIGHT TEST

## Air Vehicle Maintenance

WBS CODE: 4.41.2

#### General:

The air vehicle maintenance concept, which was established to provide the most timely and economical scheduling and the maximum utilization of resources, was as follows:

"...Flight Test to receive the air vehicles from Manufacturing and perform prior-to-flight tests at Palmdale. To accomplish this,Flight Test will use operational test procedure Technical Guides and the assistance of Manufacturing. To maintain and update the air vehicles using the guidance of NR process specifications, blueprints, Engineering Orders and Air Force Technical Orders utilizing NR assigned facilities at EAFB and AFFTC shops. To develop a series of workbooks, forms and procedures to be utilized through all maintenance cycles. To maintain all Flight Test Ground Support Equipment (FTGSE) with the assistance of NR supporting departments and the AFFTC...."

Basically the original concept was followed. The air vehicle was received and prior-to-flight tests made utilizing some Manufacturing support. A number of planned prior-to-flight operations were delayed, however, since some systems were not required for first flight. After the first few operations it was evident that changes in preliminary procedures and forms were required. Some methods of operation even dictated changes in the process specifications. Sign-off forms were constantly revised to eliminate excessive signoffs, servicing, etc., throughout the program.

The plans to utilize the AFFTC support shops to the maximum as directed by the Services & Support Agreement proved impractical in most areas and shop work ended up being performed in-house by expanding NR capabilities.

During the XB-70 Flight Test Program, specific maintenance operations were performed at specific periods and within certain constraints. These maintenance operations are defined in the following definitions:

## Flight Test Ground Support Equipment

Flight Test Ground Support Equipment (FTGSE) was defined as any or all implements or devices as approved on the FTGSE Nomenclature List required at the test site to inspect, test, service, adjust, calibrate, gage, measure, repair, assemble, disassemble, transport, safeguard actuate, and/or otherwise maintain the XB-70A Air Vehicle during the planned flight test program.

## Inspections ·

Inspections were maintenance requirements accomplished on the air vehicle or on equipment in order to determine its operating capability. These requirements included the operations necessary to adequately verify these



conditions. These requirements consisted, in part, of removing, replacing, gaining access to, testing, operating, checking, servicing, or in any manner conducting an investigation to determine the status of the equipment being inspected. Repairing, reconditioning, and unscheduled replacement were not functions of inspection, but were the results.

## Preflight Inspection

The preflight inspection was accomplished prior to flight to verify air vehicle and subsystem condition and flight readiness. The preflight inspection was divided into a hangar preflight and flight line preflight inspection. The hangar preflight inspection and the hangar postflight inspection generally overlapped when no unusual amount of unscheduled maintenance occurred between flights. The hangar preflight consisted of a minimum visual inspection, close-out of some compartments, servicing of those expendables permissible in the hangar area, and initiation of subsystem operational verification checks. The flight line preflight consisted of complete expendables servicing, specific subsystem checks associated with serviced for flight systems, final inspection and close-out of compartments left open for preflight, and final subsystem operational checks from the cockpit of the flight-ready air vehicle.

Preflight Inspection (Instrumentation)

The instrumentation preflight consisted of rechecking the instrumentation recording equipment to ascertain that no failure had occurred, and to realign and balance out the system just prior to flight.

## "Turnaround" Preflight Inspection

The "Turnaround" preflight inspection was accomplished between flights when more than one flight was scheduled on the same day. This inspection was essentially the same as the standard pre-flight inspection with the addition of some specific elements of the postflight inspection.

#### Postflight Inspection

The postflight inspection was accomplished after flight to determine the condition of the air vehicle, subsystems, engines, components and accessories. It consisted of visual inspection of exterior surfaces, compartments, subsystem installations and components, subsystem operational checks, and subsystem adjustments and calibrations required as a result of the inspection. The elements of component replacement and structural repair were unscheduled maintenance and were not included as part of the postflight inspection. The postflight inspection was divided into a flight line postflight and hangar postflight inspection. The flight line postflight inspection consisted of evaluation of in-flight discrepancies and immediately apparent problems such as leaks, structural damage, etc. This inspection was primarily geared to establishing flight line activity such as engine runs, draining of fuel, etc., that had to be accomplished prior to



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moving the air vehicle to the hangar. The hangar postflight inspection consisted of the complete visual inspection, subsystem operational checks, calibrations, adjustments, and final close-out of those compartments not required open for the preflight inspection.

#### Periodic Inspection

The periodic inspection was a thorough and searching inspection of the entire air vehicle including replacement of components as specified by the NR documents. Units were disassembled, if necessary, to ascertain the stage of wear or deterioration of parts or subassemblies. Preflight-postflight inspection, plus performance of an operational and requalifying check of all functional air vehicle systems, and accomplishment of all periodic lubrication requirements (including wheel bearing lubrications), were also performed.

#### Special Inspections

Special inspections were scheduled inspections based on subsystem requirements that cannot, due to system peculiarities, be accomplished during postflight, periodic inspections, or inspections whose frequency of inspection differs from the normal postflight and periodic frequency.

#### Maintainability

This was the quality of the combined features and characteristics of equipment design which permitted or enhanced the accomplishment of maintenance by personnel of average skills under the natural environmental conditions in which it operated.

#### Maintenance :

This was the effort required to retain material in or restore it to a serviceable condition, modifying or improving equipment in use or in storage, to meet programmed operational requirements; installation-engineering and installation of fixed communication-electronics equipment and facilities. Maintenance also included the functions of servicing, trouble-shooting, repairing, inspecting, testing, and reclaiming.

#### Organizational Maintenance

This was the maintenance performed by NR flight line personnel using flight line equipment in the performance of:

- a. Preflight, servicing, postflight and periodic aircraft inspection.
- b. Calibration of systems and removal and replacement of components.
- c. Servicing, inspection and preventative maintenance of Flight Test ground support equipment (FTGSE).



#### Field Maintenance

The maintenance performed by Flight Test personnel using shop facilities provided at the Test Site by Edwards AFB and NR for detailed inspection, testing, adjusting, calibration and repair of unserviceable parts, assemblies, sub-assemblies and components in support of organizational maintenance.

#### Depot Maintenance

The maintenance performed on assemblies, sub-assemblies and components which required industrial facilities and capabilities for complete disassembly, replacement or repair of parts, rebuilding or reassembly, and adjustment and testing which restored such items to a serviceable condition. Depot maintenance was in support of organizational and field maintenance activities.

#### Scheduled Maintenance

This maintenance was defined as the planned maintenance cycles that occurs as a result of air vehicle flight time, total operating time, or calendar time. The following inspections constituted the scheduled maintenance requirements for the XB-70A air vehicles: preflight, postflight, periodic, and special inspections.

#### Unscheduled Maintenance

Maintenance which resulted from discrepancies generated by malfunctions or deficiencies found to exist during flight, or as a result of scheduled inspection, that necessitated immediate corrective action before the air vehicle could be returned to an "in-commission" status was termed "Unscheduled Maintenance". This did not include the unpredictable maintenance generated and accomplished during a periodic inspection, but did include those configuration changes made to instrumentation installations.

#### Operating Time .

Operating time was the total operating time that a unit, component, subsystem or the air vehicle was operated.

#### Flight Time

Flight time began when the aircraft started to move forward on a take-off run and terminated when the aircraft landed. A landing was considered to have taken place when the aircraft came to a stop on the active runway or when the aircraft had turned off the active - if no stop was made. The flight time as recorded by the Tower prevailed.



Ground Time

Ground time was the total operating time that a unit, component, subsystem, or the air vehicle was operated while on the ground. This operating time did not have to be recorded in every instance.

Out of Commission (Status)

Air vehicles that were out of commission were out of commission for the following reasons:

Scheduled maintenance Unscheduled maintenance Modification or configuration change Aircraft out of commission - parts (AOCP) Engine out of commission - parts (EOCP) Aircraft not fully equipped (ANFE)

#### Maintenance Tasks:

The maintenance tasks were the requirements that were necessary to retain in or restore to a serviceable condition the air vehicle or equipment. These tasks were associated with all phases of the maintenance cycle and included such functions as servicing, troubleshooting, repairing, inspection, testing, and reclaiming. The tasks were itemized in general in a Maintenance & Support Plan. Some of the tasks outlined were combined, added to, compressed or eliminated. Thrust runs were never made on this contract and it was found that sound suppression devices were unnecessary. Servicing procedures were revised as learning progressed through the program.

Ground Handling, Towing

Towing proved to be generally easier than anticipated. A six man team was found necessary for a "bend over" inspection, walking ahead of the main and nose gear to pick up debris which would cause tire damage.

Towing from the main gear with heavy duty tugs and steering from the nose with a smaller unit proved possible, but impractical for long tows such as from the run pad to hangar.

The GSE external braking units were attached to the A/V only when the A/V was heavy with fuel. At these times, when near maximum towing weight, two heavy duty tugs were used in tandem, on the nose gear, while pulling up the grade from taxiway to hangar.

Since the hangar floor was marginal for a heavy weight vehicle it became standard practice to deflate the A/V tires prior to entering the hangar with a heavy weight A/V. This was done just outside the doors prior to entering. The tires were inflated again when the A/V was first removed from the hangar prior to towing any distance.



When the air vehicle was in the hangar a tow bar was left attached and a prime mover stationed just outside the door adjacent to the nose. There was a bridle made to tie the AICS package dolly to the main gear. In addition, the instrumentation package dolly was tied to the AICS package dolly. This provided the capability of towing the air vehicle out of the hangar in case of emergency with the packages down.

## Ground Handling, Taxi

It was found necessary to install the tire cages immediately after taxi. These were not used for a period of time but after several occurrences they were found necessary and beneficial.

#### Ground Handling, Jacking

Jacking proved to be no major problem under normal conditions, although one failure occurred in the internal structure of A/V #2 which caused the nose jack to penetrate the foreward fuselage. Subsequently a saddle arrangement was made for A/V #1 that fit the nose jack and in case of jack pad failure, would support the aircraft and distribute the load on the fuselage structure.

The original Process Specification on jacking was followed throughout. The only change required was upping the maximum jacking weight limit.

#### Ground Operations :

After acceptance of the A/V from Manufacturing at Palmdale, a series of planned operations took place in preparation for first flight. These operations were conducted by use of Operations Test Procedures which were prepared by Flight Test Operations System Engineers for use during this phase of the program. In addition to the OTP's, the Maintenance Group prepared and utilized Flight Test Technical Guides.

Due to the urgency of getting the first flight, some system checks were delayed. Only those systems that were required for the first flight were completely checked out. As the program progressed and the learning period extended, changes were made in the methods used for ground operations. These changes included short cuts to expedite the operations. In some cases, however, additional work was added which was found necessary due to experience and the OTP's and technical guides were updated accordingly.

Under normal circumstances, most operations required for postflight and/or preflight were performed in the hangar. There were exceptions to this, especially during quick turn-arounds when it was necessary to leave the A/V on the run pad. In addition, the fuel management and boost pump checks were made on the ramp or run pad after partial fuel servicing. It was found on occasion that if the A/V restraining bars were installed prior to fuel servicing that extreme difficulty could be experienced in disconnecting them. Exhibit 25, page II-429, present the XB-70 on the "run pad" in preparation for refueling.



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The quick disconnects for the hydraulic systems caused many problems during the early portion of program until reworked connectors were provided. The length of time required to disconnect was particularly obvious due to the requirement of disconnecting just prior to taxi out.

For operations away from the main facility, it was found that portable print and raw stock carts were a must. These were used for supporting Hydraulic, Electrical, Engine, Instrumentation, etc., operations.

When all engines were shut down, ground personnel could not contact the A/V crew on intercom because there was no A/V battery. To correct this situation, an AIC 17 battery box was inserted in the system between the ground crew and the A/V. It was necessary to modify the A/V to do this. The pilot selected intercom ground power and communications could be established. This modification was made only on A/V #1.

The original inert tank entry procedures were, for the most part, found to be satisfactory for inspections, minor repairs and most component replacements. When major maintenance was required in the fuel cells, it was found necessary to flush the tank or tanks with trichloroethylene and air wash them due to the time restrictions and physical restrictions of personnel suited up for inert tank entry. The only major modifications to tank entry equipment were an improved intercom system and the addition of a second emergency air supply. Fuel cell washing with trichloroethylene required revised methods to be developed from that planned. The spray bar sprinkler system of introducing trichloroethylene to the tanks for flushing worked very well for all tanks except 6, 7, and 8 which required a fill and drain procedure.

In the beginning of operations at EAFB, no facilities were provided at the Run Pad. This proved to be completely unsatisfactory. To correct this, a portable building 24 X 20 feet was brought from Palmdale by NR and installed adjacent to the Pad. This building was equipped as a field office for Maintenance and Inspection operations. The building was equipped with radio transceiver for controlling ground vehicle movement and with interphone and flight monitoring capabilities via telephone patch.

Ground cooling of the A/V was a problem due to environmental conditions at the test site. This was compounded by the hangar configuration which required long duct runs to reach the A/V. The long runs were due to placing the ground cooling carts outside the building to reduce the noise level. In addition, a sound barrier wall was necessary around the units (3 or 4 manifolded together) to cut down noise in adjacent offices.

All FTGSE equipment was serviced and maintained by the Flight Test GSE Maintenance and Control Group with assistance as required from the AFFTC and Plant Maintenance. This included fuel, oil, NH3, LN2, LOX, GN2, and dispensing equipment. Assistance was also given to the air vehicle crew for proper operation of the equipment when required. Originally the GSE group performed air vehicle ammonia servicing and assisted in other operations, e.g., refrigeration package pre-installation checkout, etc. As



time progressed and personnel became limited, all servicing was performed by air vehicle crew members.

The engine run pad and flight station were set up with all required equipment, in preparation for engine runs and flight, by the GSE group prior to moving the air vehicle to the location. This equipment was also returned to the hangar by this group after completion of these operations. Equipment located at the air vehicle was given a daily inspection by GSE Group to assure that equipment was adequately serviced and operated properly.

Since initiating the Flight Test program it was found that Flight Test needed the capability for supporting the XB-70 with GSE at sites other than EAFB. Such a requirement could be from either scheduled or unscheduled landings. To react with the necessary speed to minimize the expenditure of manhours, a fly-away kit was prepared consisting of general type equipment required for preparing the air vehicle for flight including servicing, pre and post-flight inspection and replacing tires. An alternate method was also devised for deaerating the fuel during the fueling operation. The alternate method was required due to the size and/or availability of GN2 equipment. Transporting and servicing the equipment were both problems which should be considered for this type equipment in the future.

For the most part A/V servicing was performed as outlined by the Maintenance Plan. Defueling operations required the major revision. This revision was to provide a closed loop system for all defueling and/or draining operations.

Servicing was normally performed on the Run Pad, but on occasion all expendables, except JP-6 fuel and NH3, was accomplished in the hangar. However, when this was done, generally some top-off was necessary on the pad. Exhibit 26, page II-474, present the XB-70 with FTGSE positioned during preflight in preparation for a mission. Exhibit 27, page II-431, presents the XB-70 after a mission showing the required FTGSE.

The wet ice contour molds for FACS sensor cooling had to be installed within four hours of takeoff which made it a definite pad servicing. It had also been forecasted that in-flight temperatures above 1.7 Mn would require dry ice cooling for the fire extinguisher system. This proved to be unnecessary and none was ever used. It was later estimated that anything below Mn 3 for two hours would be satisfactory without any ice cooling.

To expedite engine servicing both the hydraulic and engine oil service units were combined on one F2 trailer. This proved especially beneficial during off-base operations for handling and transportability. Engine servicing originally required the hydraulic system to be drained completely and refilled with a known amount of fluid since no quantity gage was installed. This was a time consuming operation. Later on, a fluid level checker was acquired which was a Piston Position Indicator (PPI). With this unit it was possible to locate the GN2 actuated pistons position in the reservoir thus obtaining a fluid quantity indication.



#### Maintenance Scheduling:

A procedure was established on the XB-70 program for holding a daily schedule meeting. This meeting was held approximately 1 hour after the start of the first shift. The Design Engineering Supervisor (with Systems Specialists, as required), Maintenance Supervisors from each A/V, Planner and Scheduling ECC, F. T. Operations Engineer, Inst. Engineers, Material & LSS Supervision, Inspection personnel and Air Force were always represented. In addition, the Chief Engineer and other Management personnel would attend when on site. Work required was scheduled and placed by shift on the scheduling board for each A/V. This information was copied, duplicated and distributed to all concerned by the ECC Scheduler after the meeting.

Shift turn-over notes were kept by Maintenance supervision and the status of each scheduled item noted at the end of each shift. In addition, there was a day-night Maintenance turn-over meeting which was held at the end of the first shift each day. These meetings and the notes from each shift along with the remainder of the crews' personal coordination provided adequate Maintenance continuity.

For the XB-70 program, a new group was established for configuration and accounting control and time records. One of the functions of this group was to monitor all incoming EO's, record in the Certification Verification Record (CVR) book each EO listed on the EO transmittal form, list applicable EO's in the aircraft work book and maintain active and completed records of all EO's. The IBM computer system proved very efficient since initiated for configuration and accounting control.

Time Recorders were established and required to have an accurate listing of installed controlled components at all times from Air Vehicle acceptance. To accomplish this the Time Recorder had to be in close coordination with Manufacturing and Inspection to obtain final serialization and configuration data and verify previous recorded data. This information could only be obtained and maintained with all required records which were not available from any single source. Each final installation and/or change thereafter required the incorporation of this information into the Master Card File and IBM Tab Report. In addition, each monthly revision of the Controlled Parts List Mandatory Removal and Inspection Report had to be reviewed and additional items and/or changes be incorporated into the above operation.

All spare ballistic charges and propellant actuated devices, used in the Air Vehicle crew escape system were listed on records and IBM tab in order to maintain a closer control of these units as to load date and storage life.

To minimize extended Air Vehicle lay-up's due to periodic inspections and/or replacement of controlled components, a Controlled Progressive Maintenance (CPM) program was established. The air vehicles periodic inspections and reports were divided into eight (8) CPM cycles with a lead and lag time for each cycle. Each cycle was governed by flight time and relative engineering requirements.



A complete historical file was maintained for each XJ-93 GE engine to record operating time both ground and flight. This determined time of required engine inspection and overhaul and compliance with required engine specification and EO's. This file also provided information as to engine availability and configuration. A status board was maintained in the scheduling room to provide ready, up-to-date information as to installed and spare engine status and critical air vehicle components.

#### Maintenance Shops:

The Maintenance Shops at EAFB were established to perform organizational, Field level and Limited C&E (Component and Equipment) maintenance. This was done in accordance with the Maintenance Plans. In most cases, the shops were called upon to perform more functions than had originally been scheduled.

The tubing shop, electrical shops, and filter cleaning shop had to be relocated intact from Palmdale to EAFB. This was necessary for support after Palmdale phased out. The Filter Cleaning shop was located in the M & M Hangar at EAFB instead of the NR-assigned building. Even though in another area, NR maintained Management control of the area and equipment. More and more effort was expended near the end of the program for AF support and an AF man worked in the area most of the time.

#### Maintenance Engines:

Engine maintenance was performed as a joint effort between NR shops, GE shops and with the support of the Air Force Flight Test Center (AFFTC) Propulsion Branch. All leak check, etc., runs were made in the GE test cell at EAFB prior to installation. These runs were monitored by NR Inspection in the initial stages of the program, but this was later discontinued and operations certified by GE were accepted by NR.

The working agreements between GE and NR were in accordance with the Engine Maintenance Plan. According to this agreement, NR Engine Shop performed periodic inspections, build-ups, instrumentation and removals and installations.

#### Maintenance Support:

The Flight Test Center/NR agreement was used very little in actual air vehicle maintenance. The AFFTC shops were utilized only a few times for emergency work, but primarily all work was done in NR on-site or PMD/LAX backup shops. The tire and parachute shops were probably the most extensively used support along with fire standby, supply and fueling. There were a few problems encountered in the fuel servicing but generally excellent support was given. Support in these areas was enhanced by using AF personnel assigned to the JTF for coordination.

Manufacturing and Maintenance departments in Palmdale were planned to be used as backup support for EAFB operations. These departments were utilized through the completion and first flight of A/V #2. After this time, shops that had been



anticipated to remain in Palmdale were closed down. This closing necessitated the acquisition of equipment and skills by Flight Test and the relocation of complete shops to EAFB. This was accomplished by what was termed the FLIGHTS Committee (Flight Test Support EAFB), which was made up of Manufacturing, Industrial Engineering and Flight Test personnel.

The complete tubing shop, filter cleaning and electrical maintenance shops were relocated to EAFB when Site #3 phased out at Palmdale. The electronics lab functions were taken over by Flight Test and the electronics shop expended and equipped to perform the work that had been handled previously in Palmdale.

Shop support, material follow-up and Logistics supply functions in Los Angeles were utilized on the program. The systems functioned as planned and procedures appeared to be adequate in this area.

General Electric provided engine support with ample representation at Palmdale, Santa Susana, LA, AEDC, and EAFB. The GE shops at Edwards and Evendale were utilized in support of this program. A working agreement was negotiated to define the areas of responsibility between GE and NR.

#### Maintenance Inspection:

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Air Force Quality Assurance representatives had coverage on all shifts during the entire program. The A/V's were accepted by the AF by a 262 inspection prior to first flight.

Flight Test Maintenance personnel prepared the various inspection forms which were then approved by Quality Control. These forms were continually revised during the program as experience was gained. The forms initially were compiled into six books for each A/V. These consisted of the following:

Book #1	Carried the Work Items
Book #2	Parts Replacement Records
Book #3	Post Flight
Book #4	Preflight
Book #5	Instrumentation
Book #6	Flight Records

As time progressed and a number of fuel leaks occurred, a Book #7 was added to maintain records of these leaks. In addition, a Book #8 was added for Mandatory Inspections.

An on-site fluid test lab was established to provide analysis of fuel, hydraulic oils, and other expendables. Both Air Force and Company inspectors were assigned to cover shipping and receiving of parts and components at the site. In addition, inspection coverage was provided for all Facilities and FTGSE equipment maintenance and repair.

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#### Communications:

The ground communications network for the XB-70 Flight Test Program was broken down into the following categories: Data Control, Voicecom Center, Flight Shack, Mobilcom Bus, radio equipped vehicles, portable radios, and ground maintenance intercom. Each category is analyzed as to its original purpose, problems involved in meeting initial requirements, actual use during various phases of the program, capabilities added and requirements deleted.

#### Data Control

Edwards Data Control, at Building 3940, was designed as a flight control center for the NR Flight Test engineers during flight. Telemetry parameters were displayed on plotting boards or oscillographs as required. Closed circuit television was available for observing aircraft up to a distance of one hundred and fifty miles. An extensive communications network was maintained to permit contact with the air vehicle, the ground maintenance crew, and the engineering staff.

Two UHF frequencies were available for maintaining contact with the air vehicle and its chase aircraft. The two radios used were selected from a bank of four locally operated GRC-27's, or from a bank of remotely controlled GRC-27's located at Building 2580, or from "High Range" radio stations remotely controlled by land lines or micro wave. Initially all radios used at EAFB were remotely controlled from Bldg. 2580. Due to the unreliability of this system, the four GRC-27's were installed in Building 3940. This arrangement proved to be satisfactory. The "High Range" network was initially subject to numerous equipment malfunctions; however, as the program progressed, communications became more reliable. Many of the malfunctions later consisted of loss of reception by the XB-70. During these periods excellent reception of XB-70 transmissions at Data Control usually prevailed. Use of chase aircraft for relay stations was used extensively when the chase aircraft were properly positioned.

Three Gonset 150 radios were initially installed at Edwards Data Control. One unit was tuned to 123.15 MC for cummunications with the ground maintenance vehicles. Another was tuned to 123.35 MC for cummunications with Flight Test Operations vehicles. The third unit was tuned to 123.5 MC which was the transport aircraft dispatch frequency. Due to the lack of use, the 123.35 MC radio was removed from operation. The 123.15 MC and 123.5 MC radios were not used extensively but when needed both proved to be indispensable. Any time aircraft carrying only VHF equipment were involved in XB-70 tests, 123.15 MC was used as the primary VHF frequency to avoid cluttering up 123.5 MC which was used by other aircraft companies. The 3940 Data Control equipment and the P.A. system was maintained by AFFTC personnel. NR employees maintained the Contractor-furnished Gonset radios used in Data Control as well as other ground communications utilized in the local areas. This maintenance was performed in Building 1820 Electronics Shop.



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At the beginning of the XB-70 program an elaborate thirteen channel intercom system was used. The malfunction rate was high and it was eventually replaced by a GTA-6 telephone system. The GTA-6 was less versatile and the intercom capabilities were not used extensively. Due to the close proximity of all personnel concerned in Data Control, direct verbal communication proved to be more satisfactory than using the intercom system.

Two hours prior to flight, a one-way intercom phone patch was set up from the Mobilcom Bus at the engine Run Pad to Data Control. This patch was used to keep the engineering personnel informed as to the status of the air vehicle.

Two telephone lines were present between Building 1820 and 3940 to distribute the local and "High Range" UHF audio to key points throughout Building 1820. This audio also went to the PBX room and could, by use of telephone lines, be routed to the Flight Shack, Security Shack and Los Angeles. The primary UHF frequency was normally sent on one of the lines from Building 3940 and secondary chase frequency was transmitted on the second line. Several telephones were required to maintain contact with necessary personnel during flight. One NR line, three base lines, a hot line to the Control Tower, and a hot line to RAPCON were available and were used constantly during flight.

#### Building 1820

The communications network at the XB-70 Flight Test Facility, Building 1820, was built-up around the Voicecom Center. The Voicecom Center was equipped with three UHF radios, two VHF radios, one HF SSB radio, an intercom to the electronic shop, a radio phone patch, two telephones, and two tape recorders. The Voicecom Center was originally designed to coordinate all mobile vehicle movements and as a backup communications center in case of radio failure at Data Control. Due to the cutback of the XB-70 Program at the time A/V #3 was cancelled, the Voicecom Center never was used to the extent of its design. It was used during flight program as a reception point for messages to occupants of the building from mobile vehicles operating remotely, and recording flight data. Because of its high antennas it proved itself invaluable as a relay station for vehicles out of range of each other.

Three GRC-27 UHF transceivers located in the electronics shop were remotely controlled from the Voicecom Center. These radios were used to check out the entire UHF communications network prior to each flight. During flight they were used to monitor the flight and would have been used to control the mission if necessary.

Two Gonset 150 VHF transceivers were available for communicating with NR transport aircraft (123.5 MC) and maintenance personnel (123.15MC). During emergencies these radios played a key role in dispatching GSE to required areas and relaying messages from the Mobilcom Bus and other mobile vehicles to the necessary personnel in Building 1820. A public address system was at the disposal of the Voicecom operator allowing him to page any one, or any combination of twelve areas around Building 1820.



A Collins, HF SSB, KWM-2 transceiver was located in the electronics shop and was remotely controlled from the Voicecom Center. This unit had a power output of 100 watts PEP and was designed for long range communications. The two frequencies which had been assigned to NR by the FCC were 3376KC and 6770KC. These frequencies were shared with other contractors on base. The intended purpose for this unit was for long range communications in case the XB-70 landed in some remote area. A second KWM-2 was located in the Mobilcom Bus. Experience has shown that remote control of a unit such as the KWM-2 was impractical because of the frequent tuning requirements. The KWM-2 was never used as designed and during tests was operated locally from the electronics shop.

An electronic audio indicating system was used in the Voicecom Center to give the operator visual indication of the source of his audio signal. This indicating system eliminated much confusion.

Two voice operated Ampex tape recorders were located in the Voicecom Center to record the XB-70 flights. The signal recorded originated at Building 3940 and included the "High Range" audio. The recorders were originally located in the engineering office, but were removed because of the distraction caused during playback and the requirement for constant monitoring during flight. VOX operation was utilized to conserve tape and **caused a certain loss of audio** at the beginning of each series of transmissions. With a good signal this loss was usually negligible.

Two telephones were used in the Voicecom Center. Originally there was also a hot line to Edwards Control Tower. This line was later disconnected because of lack of use. The importance of an adequate telephone system in an operations center of this type cannot be over emphasized.

A five-channel radio telephone patch was in operation in the Voicecom Center to permit radio communications from any UHF, VHF, or HF station within range of the Voicecom Center to any telephone capable of reaching the NR operator. Since the system was voice operated the success of the patch was dependent upon a good signal to noise ratio on the phone line. This was always obtained on local NR lines but seldom on incoming calls from outside lines. (A more reliable arrangement would be a manually operated system.) The phone patch was used extensively as a one-way patch (no talking capabilities) out of Building 1820. (In the past the X-15 flights were monitored in Los Angeles using this method.) In addition to the five two-way radio telephone patches available from the NR telephone operator, there was also a one-way radio phone patch which carried the XB-70 local and "High Range" audio. This was the same audio line which was connected to the tape recorders. It originated at Edwards Data Control.

Flight monitor speakers were located in the Flight Debriefing Room, Flight Monitor Room #3 and #4, Project Engineer's Office, Electronics Shop and Voicecom Center. During a flight these speakers carried the local and "High Range" audio from Edwards Data Control. Two hours prior to each flight, the XB-70 interphone audio was connected into the Flight Monitor Rooms #3 and #4 via a conference call phone patch from the Mobilcom.

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At the beginning of the program, interphone stations were located in Flight Monitor Rooms #3 and #4. These two stations were a part of the interphone system from Data Control. When the GTA-6 system was installed the two stations were eliminated due to lack of use.

#### Flight Shack

The NR Flight Shack was located at the XB-70 engine Run Up pad. It was utilized by the XB-70 ground crew as a supervision and inspection office during the times the XB-70 was located on the Pad. A Gonset 150 transceiver, tuned to 123.15 MC was present for communications with the Mobilcom and other mobile vehicles during flight. With the help of this radio, air vehicle GSE could be dispatched to the runway or any other place required in a minimum of time. A 60-watt PA system was utilized for paging purposes during operations in the area. During flight the local and "High Range" audio was piped into the flight shack from the NR PBX office and presented over intercom and PA system. This kept the ground crew informed as to the progress of the flight.

Provisions were available at the Run Pad for the one-way AIC-17, intercom phone patch to Security, Flight Monitor Rooms #3 and #4, Edwards Data Control and Los Angeles. This patch was set up from the Mobilcom and was used to keep all areas up to date on prior to flight operations.

The AIC-17 interphone power supply, three AIC-17 interphone stations and an intercom monitor speaker were available in the Flight Shack for use by inspection and the XB-70 ground crew during operations.

#### Mobilcom

The Mobilcom Bus was the operations center for the XB-70 at all times the air vehicle was on the ground but not in the hangar. It was able to operate from a fixed location or in a mobile configuration. During preflight operations at the engine Run Pad the ground crew utilized the Mobilcom as a control center for coordinating air vehicle maintenance. During engine starts prior to flight, operations engineers used the Mobilcom as a control center for checking operation of the air vehicle systems. While taxiing, the ground crew escorted the air vehicle with the Mobilcom advising it of any problems and assisting the pilots as necessary. During ground emergencies (such as blown tires on the runway) the Mobilcom was utilized as a communications center. Messages were relayed to and from Data Control, Voicecom, and the Flight Shack concerning GSE requirements and other information pertinent to the situation. If any part of the runway or taxiway was blocked, contact was established with Edwards Ground Control giving them frequent progress reports.

The Mobilcom was equipped with two ARC-27 UHF transceivers; one Gonset 3139 VHF transceiver, one Gonset 3156 receiver, a five-channel intercom, a 75-watt PA system, a tape recorder, and had instruments to indicate wind temperature and altimeter information.



The ARC-27's were used for UHF communications with the XB-70 Data Control, Voicecom, or the tower. The ARC-27's could be tuned to any frequency between 225.0 and 339.9 MC and had a power output of 15 watts.

The Gonset 3139 was used for VHF communications with Data Control, Voicecom, Flight Shack or other mobile vehicles. The transmitter was crystal controlled and could be tuned to any one of six frequencies, with a power output of 6 watts. It was normally tuned to 123.15 MC during flight. The other crystals were for 123.35 and 123.5 MC.

A Gonset 3156 receiver was available for listening to any station in the frequency range of 108-135 MC. It was seldom used.

The tape recorder in the Mobilcom was operated continuously or in a VOX mode. It was capable of recording any one of the radios or intercom channels. The recorder was operated at all times that any operation was in progress. Only the tapes concerned with flights or special tests were retained.

Two telephones were available in the Mobilcom. One was a base line and the other an NR line. These were attached by means of phone jacks mounted on the bus. A one-way phone patch was available in the Mobilcom. Two hours prior to each flight it was utilized to send the AIC-17 intercom audio to Data Control, Security, Flight Monitor Rooms #3 and #4, and Los Angeles.

Metro Bus

The Metro Bus was originally built up for the X-15 Program. It provided many of the capabilities that are found in the Mobilcom; however, it lacked the room and was used as a backup vehicle. The Metro Bus contained provisions for one operator only. Five or six persons in addition to the operator could be seated in the Metro, but they were without desk space. Moving from one place to another in the Metro was virtually impossible and no air conditioning was available. The Metro Bus was used quite frequently as an operations center when both aircrafts were out of the hangar simultaneously. On occasion it was utilized when the Mobilcom was out of commission.

The Metro contained two ARC-27 transceivers, one Gonset 3139, VHF transceivers, one Gonset 3139, VHF transceiver, provisions for a KWM-2 transceiver (the unit which could be placed in the Metro was in the Mobilcom), a PA system, a tape recorder and two telephones. When the Metro Bus was utilized for flight, the two ARD-27's were used for UHF communications with the XB-70, Data Control, Voicecom, or the Tower and they could be tuned to any frequency between 225.0 and 399.9 MC with 15 watts power output.

The Gonset 3139 was used for VHF communications with Data Control, Voicecom, Flight Shack, or other mobile vehicles. The transmitter was crystal controlled and could be tuned to any one of six frequencies, provided all six crystals positions were utilized, with 6 watts of RF power output. Normally this unit was tuned to 123.15 MC.

The KWM-2, when utilized, was operated on either of the two Company frequencies of 3376 IC and 6770 KC.

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A public address system was available for paging purposes and for broadcasting the UHF audio.

The tape recorder in the Metro could tape only the UHF audio and VOX operation was not available. Wind, temperature and altimeter indicators were also available.

#### Miscellaneous

Two jeeps were used during the program. One jeep carried an ARC-27 UHF transceiver and a Gonset 150 single-channel transceiver tuned to 123.15 MC. The ARC-27 used a 24V aircraft battery for power. The battery was normally charged with a trickle charger. A 24VDC gas driven generator was mounted in the jeep but was too noisy for continuous operation. A more satisfactory arrangement would be a 24V alternator running off the jeep engine. A fully charged battery was capable of delivering adequate power for about one hour. The UHF was used primarily for maintaining contact with Ground Control while the XB-70 was being towed to and from the engine run area. It was also used to maintain contact with the tower during operations on the lake bed or other areas requiring Ground Control coordination.

The VHF was used primarily during emergencies for maintaining contact with the voicecom of the Flight Shack while the jeep was used to coordinate movement of GSE items.

The other jeep had a single-channel Gonset 150 transceiver, tuned to 123.15 MC. The vehicle was used for general transportation and/or support. The radio was primarily used during aircraft emergencies to dispatch GSE.

The personnel carrier contained a single-channel Gonset 150 transceiver tuned to 123.15 MC. It was used to transport the maintenance crew during emergencies.

The Dodge Power Wagon contained a single-channel Gonset 150 tuned to 123.15 MC. This unit was used to deliver GSE during the XB-70 operations away from the hangar. The radio was used primarily during emergencies for coordinating GSE movements. The original task for this vehicle was to retrieve the deceleration chute after A/V landing. The AFFTC performed this task.

The Air Vehicle Tow Tractors both had provisions for a Gonset 150 VHF transceiver tuned to 123.15 MC. One radio was available and was swapped back and forth as necessary. The Tow Tractors were used to tow the XB-70 where required. The radios were used to keep in touch with the jeep which was in turn to contact with the tower on UHF.

Two station wagons were equipped with radios. One with provisions for a Gonset 3139 transceiver. The radio was removed from the vehicle due to lack of use. The other station wagon was equipped with a Gonset 150 VHF trans-ceiver tuned to 123.15 MC. The radio was often used while towing the XB-70 to keep in touch with the jeep and the tow tractor.



Four portable battery operated Skycrafter transceivers were used on various occasions during the XB-70 Program. They were single-channel units with an output of one-half watt of power. The units were very reliable for distances under a quarter of a mile. At greater distances antenna placement was quite critical and untrained people experienced a great deal of difficulty with them. Reliable air to ground communications was consistantly obtained up to three miles.

The AIC-17 intercom system was used by the ground crew for communications between the air vehicle, the ground crew, the Flight Shack and the Mobilcom. Any number of stations could be added to the system without loading or other adverse effects. A malfunction at one station would not affect the rest of the network. Instead of using the PP1618/AIC-17 battery packs, a separate 24V power supply was manufactured. This power supply eliminated failures due to dead batteries. Interconnected junction boxes were placed at strategic points around the air vehicle so the ground crew could plug into the necessary box with extension cords.

#### Long Line

Long Line communications are ground-to-ground communications such as TWX, telephone, etc. The AFFTC facility was provided an in-plant dial system and a manual PBX board by the AFFTC. The board was manned by NR operators for two shifts. The telephone system was equipped with a master intercom connecting the major offices. In addition to the master, each system had a local intercom from the secretary or clerk to each instrument. Each functional group had a three or four number rotary which made up their system. Under the initial Joint Test Force operation the systems were layed out to integrate the AF personnel with the contractor counterparts. This proved unsatisfactory and had to be reworked to provide the functional group systems.

The NR EAFB telephone system was connected to the Palmdale and LA boards by lease lines. The NR board at EAFB was linked to the AFFTC main switchboard with local trunks and in addition some of the lease lines to LA and Palmdale were terminated on the AFFTC board. This termination provided flexibility for on-base calls and coverage during off hours. For emergency call-ins and night lines, both Palmdale and Lancaster FEX trunk lines were terminated on the NR/EAFB board. There was also a radio telephone "patch system" installed which provided listening capabilities for flight operations at selected speaker phone locations.

An Engineering and Logistics supply TWX system was installed linking the EAFB Facility with the corresponding offices in LA and Palmdale. One machine was located in the Engineering area at EAFB and one in the Material Control Office. One additional TWX circuit was installed and this provided pilot weather information and was located in the Dispatch Office.

A complete PA system was installed which included links to all the major Base facilities utilized, such as Run Pad, taxiway 5, thrust stand, etc. These links could be selected on an individual basis from Voicecom or all positions simultaneously. Building 3940 Control Room or Security had an



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all-position override in case of emergency. In addition, selected areas had individual paging capabilities such as the warehouse area. This network was never utilized as originally planned.

North American Rockwell funded for all TWX circuits, lease lines and FEX trunks. All other services and the equipment installations were supplied by the AFFTC on a no-charge-for-use basis.

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# ORIGINAL PAGE BLACK AND WHITE PHOTOGRAPH



FTGSE - MISSION START



FTGSE - MISSION COMPLETION

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EXHIBIT

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 02 AIR VEHICLE MAINTENANCE

		DESIGN	
		/ENGR	TOTAL
		HOURS	HOURS
		DOLLARS	DOLLARS
DESIGN/ENGINEERING		219064	219064
LABOR AT \$ 4.352		953328	953328
ENGR BURDEN AT \$	5.328	1167125	1167125
SHOP SUPPORT		148211	148211
LABOR AT \$ 4.440		658064	658064
TEST/QC		64799	64799
LABOR AT \$ 4.618	• *	299264	299264
MFG BURDEN AT \$	5.746	1224034	1224034
ENGR MATERIAL		698057	698057
MPC		102548	102548
OTHER COST		10528	10528
SUB-TOTAL		5112948	5112948
GEN & ADMIN		103066	103066
TOTAL COST		5216014	5216014

TIME PHASED COST DETAIL - SEE PAGE II-433

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESIGN/ENGINEERING	
4-SYSTEM	4	
5-SUB SYSTEM	41 ATR VEHTCLE MATNEENANCE	
6-MAJ ASSY	02	
SUBD OF WORK	DESIGN/ENGINEERING	

## ON-SITE LABOR

		MAN- MON THS	LABOR HOUR S	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1 Q-2	65 65	163.5	28260	5.237	148001	180024	328025
Q-3	65 65	249.0	<b>41</b> 808	4.698	196432	236690	433122
Q-1 0-2	66 66	358.5	62159	4.403	273703	365756	639459
Q-3	66	517.5	86837	3.860	335192	384655	719847
тот	AL	1288.5	219064		953328	1167125	21 20 45 3

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 4 5-SUBSYSTEM 41 AIR VEHICLE MAINTENANCE 6-MAJ ASSY 02 SUBD OF WORK DESIGN/ENGINEERING

## ON-SITE LABOR

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-3 Q-4	64 64	1.5	206	3.981	820	1223	2043
Q-1 Q-2	65 65	85.5	14878	3.638	54130	109616	163746
Q-3 Q-4	65 65	150.0	25224	3.701	93344	186554	2 <b>7</b> 98 <b>9</b> 8
Q-1 Q-2	66 65	355.5	61535	3.579	220236	457359	677595
Q-3	66	276.0	46368	6.244	289534	469282	758816
TOT	AL	868.5	148211		658064	1224034	1882098

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC4-SYSTEM45-SUBSYSTEM41AIR VEHICLE MAINTENANCE6-MAJ ASSY02SUBD OF WURKDESIGN/ENGINEERING

## ON-SITE LABOR

LABOR + BURDEN \$	BUR DEN DOLL ARS	LABOR DOLLARS	LABOR RATE	LABOR HOUR S	MAN- MON THS		
170		170	3.208	53		-3 64	)-3
36818		36818	4.365	8435	48.0	-4 64 -1 65	2−4 2−1
63491		63491	4.502	14103	84.0	-2 65 -3 65	2−2 2−3
112892		112892	4.801	23516	135.0	-4 65 -1 66	)-4 )-1
85893		85893	4.595	18692	111.0	-2 66 -3 66	2−2 2−3
299264		299264		64799	378.0	TOTAL	TO

## APRIL 1972

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 02

AIR VEHICLE MAINTENANCE

AIR VEHICLE MAINTENANCE

		MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
Q-3 Q-4	64 64	1.5	259	3.822	990	1223	2213	127
Q-1 Q-2	65 65	297.0	51573	4.633	238949	289640	528589	26934
Q-3 Q-4	65 65	483.0	81135	4.354	353267	423244	776511	19976 <b>7</b>
Q - 1 Q - 2	66 66	849.0	147210	4.122	606831	823115	1429946	204436
Q-3	66	904.5	151897	4.678	710619	853937	1564556	266793
TOT	AL	2535.0	432074		1910656	2391159	4301815	69805 <b>7</b>

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 AI 6-MAJ ASSY 02 AIR VEHICLE MAINTENANCE

AIR VEHICLE MAINTENANCE

MFG TOTAL OTHER SUB TOTAL MATL MATERIAL MPC COST TOTAL G & A COST Q-3 64 127 46 2386 51 2437 Q-4 64 Q-1 65 26934 8056 1349 15072 564928 580000 Q-2 65 Q - 3 65199767 35638 981 1012897 27024 1039921 Q-4 65 Q-1 66 204436 42400 2012 1678794 5084 1683878 Q-2 66 Q-3 66 266793 16408 6186 1853943 55835 1909778 TOTAL 698057 10528 5112948 103066 102548 5216014



#### WBS 4.0 FLIGHT TEST

#### INSTRUMENTATION

#### WBS CODE: 4.41.3

This section presents a summary of the flight test instrumentation operations. For description of the Instrumentation Subsystem and associated exhibits see WBS 1.11 in Volume IV of this report. This summary discusses the flight test instrumentation operations under six major functions: Pre- and Post-Flight Procedures, Calibrations, Maintenance and Improvements, Data Acquisitioning, Data Accuracy, and Reliability.

#### Pre- and Post-Flight Procedures:

The Instrumentation Pre-Flight Procedure was divided into three phases and were referred to as T-24, T-9, and T-3. Each of these phases were defined in detail in Flight Test Technical Guide Reports (FTTG) as listed below:

- (a) T-24: Instrumentation System & Sensor Operational Status Checks.
- (b) T-9: Instrumentation Pre-Flight Data Acquisition.
- (c) T-3: Instrumentation Flight Configuration and Clearance Procedure

The above listed procedures were reviewed and revised at various times to reflect any new or improved methods that would improve data accuracy, reliability, and maintenance.

#### T-24 Procedure

The T-24 Instrumentation Procedure was initiated as early as possible before flight to insure that all systems and sensors were operating properly prior to taking formal pre-flight data records. An outline of this procedure follows:

#### Instrumentation Operational Status Checks:

- (1) Package Configuration
- (2) Instrumentation Control Switch Configuration
- (3) Circuit Breaker Configuration
- (4) Decommutator Selector
- (5) Package Cooling(6) Application of I
- (6) Application of Package Electrical Power
- (7) Digital Data System Checkout
- (8) Digital Parameter Checkout
- (9) Analog Data System Checkout
- (10) Telemetry Transmitting Equipment
- (11) Digital Recorder Checkout
- (12) Analog and Telemetered Parameter Checkout
- (13) Environmental System Checkout
- (14) Cockpit and Cabin Equipment Checks
- (15) Flutter Recording System



- (16) Landing Gear Camera Checks
- (17) SST Recording System Checkout
- (18) Strain Gauge Power Supply Resistance Check
- (19) CCC Pressure Transducer Power Supply Frequency and Voltage Check
- (20) Analog System Oscillator Frequency Check

#### T-9 Procedure

The T-9 Instrumentation Procedure was initiated after completion of the T-24 Procedure, and as close to flight time as was possible. This procedure, when followed in its entirety, assured data reduction accurate pre-flight reference and sensitivity levels for all sensors. Upon completion of the pre-flight data records, the data tapes and all necessary information, including a copy of the T-9 Data Log Sheets were shipped to L.A. Data Reduction by the fastest possible transportation. An outline of this procedure follows:

#### Analog and Digital Record #1

(1)	A/V Circuit Breaker Configuration
(2)	A/V and Package Power Requirements
(3)	Instrumentation Package Circuit Breaker Configuration
(4)	Tape I.D. Switch Configuration
(5)	Time Interval Record Switch Configuration
(6)	Instrumentation Package Control Switch Configuration
(7 & 8)	External Equipment Requirements
(9)	Cabin P.D.U. Circuit Breaker Configuration
(10)	Cockpit Control Switch Configuration
(11 & 12)	A/V External Jig Requirements
(13 thru 20)	General Information Prior to Taking First Digital Data Record
(21 thru 23)	Digital Data Record #1
(24 thru 25)	General Information Prior to Taking First Analog Data Record
(26)	Analog Data Record #1

Analog and Digital Record #2

(1)	A/V Power Requirement
(2)	A/V Circuit Breaker Requirement
(3 thru 7)	General Information Prior to Taking Digital Record #2
(8)	Digital Data Record #2
(9 thru 10)	Analog Data Record #2
(11 and 12)	Removal of A/V Jigs and Equipment

#### Analog and Digital Record #3

(1)	Cockpit Instrumentation Con	ntrol Switch configurati	on
(2)	Instrumentation Package Cor	itrol Switch configurati	on
(3)	Digital Data Record #3	0	



#### Digital Data Record #4

(1 thru 3) Preparation for Digital Record #4
(4) Digital Data Record #4

Pilots Vertical Accelerometer Check and Instrumentation Package Power Shut Down

Log Sheets for Data Record #1 (4 Sheets)

Log Sheets for Data Record #2 (2 Sheets)

T-3 Procedure

After completion of the formal pre-flight data records, the instrumentation package and instrumentation section of the AICS package were inspected and readied for flight. The T-3 Procedure became a permanent record and sign-off sheet to verify that the package's modules, control panels, circuit breaker panel, tape recorders, and so forth, were in the proper configuration for flight. This procedure, when completed, was the final step in the instrumentation system for pre-flight. An outline of this procedure follows:

(1)	Camera Close-Out	
<b>(</b> 2)	Pilots Instrumentation Control Panel	
(3)	Co-Pilots Instrumentation Control Panel	
(4)	Pilot and Co-Pilots Correlation Panel	
(5)	Pilot and Co-Pilot Instrument Panel (Instrumentation)	
(6)	E.E. Bay Circuit Breaker Configuration	
(7)	Cabin Area, Instrumentation and SST Shelf	
(8)	Data Recording System	
(9)	Instrumentation Package Circuit Breaker Panel	
(10)	Instrumentation Package Control Panel	
(11)	AICS Package (Instrumentation Sections)	
(12)	Instrumentation Power Shut Down	
(13)	Instrumentation Package Closeout	
(14)	Package Installation in A/V	
(15)	Final Instrumentation System Checkout	
(16)	Final Package L/N <sub>2</sub> Top Off	
(17)	Enclosure: Instrumentation Flight Clearance Sign Off Record	

Post Flight Procedure

In addition to the pre-flight instrumentation procedures listed above, a short informal post flight procedure was followed to insure that all data taken during a flight was properly removed, identified and shipped to L.A. Data Reduction with a minimum of time. This post flight procedure also covered the final shut down of instrumentation power.



## Instrumentation Calibrations:

To insure the validity of flight test recorded data, periodic calibrations were performed on all data sensors and associated recording systems. The overall method and interval of these calibrations for airborne data sensors were established by an NR Engineering Flight Test Procedure Letter which included a special calibration interval for each type of data sensor used in the XB-70A Air Vehicle. The recording systems calibrations were normally a function of maintenance procedures covered elsewhere in this section, principally in the T-24 pre-flight status checks. A description of each category of airborne data sensor calibration with sample procedures and forms follow. The calibrations were generally groups as those performed in the air vehicle with the data system, and those performed with the airborne sensors removed in outside laboratory facilities. In either case, it was this calibration which determined the value of the factors, in engineering units, used to process flight data.

Calibrations Performed On The Air Vehicle (Using The Airborne Data System)

Position transmitters and many miscellaneous parameters were calibrated by providing controlled inputs from A/V components. For example, elevons were deflected in known (measured) increments of full travel or the Central Air Data System (CADS) was provided simulated, precise pressure and temperature inputs while instrumentation pick offs were monitored at these known system data levels. These data were plotted to verify a linear output of the data sensors, and to determine the factor for data reduction. A calibration procedure used for elevon positions and a sample blank "universal" data sheet used for most calibrations of this type are included (Encl. 2 and 3). Calibration procedures of most all types of instrumentation used on the XB-70A are on file with the Data Acquisition Engineering Office.

Simulated inputs from portable laboratory equipment were used to verify the calibration of the signal conditioning and data systems for such parameters as thermocouples, liquid volumetric flow rate and RPM. With the data sensor disconnected, an appropriate millivolt supply source or signal generator was substituted to provide precise incremental voltage or frequency inputs to the signal conditioning and data systems. The output was then recorded at these known levels or visually observed on digital or analog readout equipment. Then tabulated data was plotted, or recorded data reduced through normal data reduction procedures, to verify the required calibration.

## Calibrations Performed Under Laboratory Conditions

Instrumentation data sensors were calibrated in the laboratory, at least initially, at several different temperature environments simulating those to be encountered in their application. Subsequent calibration of position transmitters usually consisted of those described above which were intended primarily to verify that the mechanical linkages and signal conditioning still produced a linear output at the desired level of signal. Other types of transducers, however, received a complete periodic laboratory calibration since the physical aspect of their installation was not a primary factor in the overall data sensing system accuracy, i.e., pressure transducers,

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accelerometers, gyros, etc. Also, the necessary conditions for their calibration were more easily controlled in the laboratory.

The Litton pressure transducer calibrations were verified by the Los Angeles Metrology Laboratory, normally every 30 days. When calibrated units were reinstalled, only a pressure check of the system in which they were installed was performed to insure an absence of leaks. Leakage rates were observed by monitoring the digital readout count, with respect to time, while the system involved was pressurized and disconnected from the pressure source. The high degree of accuracy of the Litton transducers, and the fact that they were used to measure basic reference parameters was justification for their frequent and precise calibration.

Other transducers which sense either pressure, or acceleration forces were usually returned to the Flight Test Laboratory for calibration at no more than 6 month intervals. However, due to their inaccessability and the large number involved, verification of previous calibrations of some pressure transducers in the air vehicle was allowed through the data system when scheduling considerations justified this action. This was accomplished by accurately pressurizing the reference systems individually, to which most were connected, at various levels and recording the outputs of a large number of transducers simultaneously.

Flow transmitters used in both the fuel and hydraulic systems were sent to the A. F. Fuel Laboratory at Edwards Air Force Base for calibrations at intervals deemed necessary. Data provided by this laboratory were then plotted to arrive at the factors needed to interpret the flight data.

The calibration of all vibration, flutter pick-ups, and boundary layer or noise level microphones was accomplished by the NR Structural Dynamics Laboratory at intervals dictated by the flight program and installation considerations. For instance, all engine vibration pick-ups were replaced with recently calibrated units at the time of every engine change. Other pick-ups were replaced with newly calibrated units as often as each flight, dependent on future flight objectives or past flight data evaluation. All pick-ups were removed for calibration during any extended layup of the air vehicle.

Rate and attitude gyros with some of the associated signal conditioning circuitry were normally sent to the Flight Test Laboratory, where rate and tilt tables were available, for periodic calibration. However, the attitude gyros were calibrated using the air vehicle data system by use of extension cables on the gyro module, and reading the output at various angles of inclination as measured with a calibrated inclinometer.

There were some special recorders installed in the XB-70 such as the NASA Airspeed Recorder, VGH Recorder and Angular Accelerometers. These were self contained sensing and recording systems which were maintained and furnished by NASA. Any calibrations required were performed by NASA during regular maintenance procedures initiated by them.



## Instrumentation System Maintenance and Improvements:

The size and scope of the data acquisition system required a continuous effort towards system maintenance and improvement in order to insure that all systems were in perfect working order and that all acquired **data** was reliable and accurate. Most of the electronic equipment was of solid state design, with very few exceptions, and for that reason alone was highly reliable and relatively trouble free. Many components were interchangeable and therefore presented a very desirable and efficient means of trouble-shooting the system, in the event of a malfunction, by means of substitution and interchanging of components. Exhibit 28, page II-447, presents a view of the instrumentation package during an "up-grading".

#### Maintenance

One of the primary means of maintenance of the entire data acquisition system was the accomplishment of the Instrumentation Pre-Flight Procedure (FTTG), prior to each flight. The procedure outlined many maintenance procedures, and system and parameter checkout procedures, and when followed to completion, did insure that all systems and data parameters were operating at their most efficient, accurate, and trouble free manner.

Another phase of system maintenance was the re-calibration of all data parameters. The recalibration period of data parameters varied from a period of one month to six months, depending on the type of transducer used. This period of recalibration was determined through past performance and reliability of various types of pickups and transducers. The "Instrumentation Calibration Procedure" part of this report defines the re-calibration period. In certain cases, when necessary due to a checkout desired to verify the proper operation of a parameter, a parameter was re-calibrated within the minimum re-calibration period. During all calibrations, and after a final analysis of the calibration data, if any discrepancies were noted that would show a variation from a past calibration, that parameter or system was carefully analyzed to determine the cause, and all necessary action was taken to correct and prevent the re-occurrence of that discrepancy.

The Data Processing Group was another very valuable source of information regarding the condition or degradation of accuracy or reliability of the data system or parameters. The Data Processing Group constantly monitored all data and informed the Data Accuisition Group of the discrepancies detected. Upon receipt of a data squawk, the instrumentation engineer analyzed the squawk to determine the possible cause and necessary action to correct and prevent further squawks of that type. In many cases the engineer determined, from records of past histories of similar squawks or from his own past experience, exactly what action was required. A written record was made and filed of all system or parameter discrepancies and the action taken to correct and prevent any re-occurrence.

Trouble shooting to determine an answer to a malfunction or discrepancy to the **data system** or a data parameter was greatly simplified, and a minimum of time was required due to the interchangeability of many of the components of the data system. In addition, the data system was divided into major areas to assist in determining an answer to the malfunction or discrepancy.

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The following shows the many varied and useful means that were used to find a solution to a problem.

#### Parameter Patch Panel

By removing patch panel the data system was divided into the following two major areas:

- a. The basic data recording system including data filters, data switches, data amplifiers, subcarriers, etc.
- b. The data input transducer including wiring, plugs, etc.

#### Interchangeable Components

- a. All 5 to 1 data switches
- b. All 4 to 1 data switches (in many cases a and b were interchanged).
- c. All signal conditioning cards of a given type
- d. All 12V power supplies
- e. All 5V power supplies
- f. All CCC pressure transducer power supplies
- g. All CCC transducer modules
- h. All 100 to 1 switch modules
- i. All 20 to 1 switch modules
- j. All 10 to 1 switch modules
- k. All 4 to 1 switch modules
- 1. All thermocouple switch modules
- m. All filter modules of a given type (when necessary all l cps, 4 cps, or 8 cps filter modules were interchanged to assist in determining a solution).
- n. All digital data amplifiers (5 total)
- o. All analog subcarrier oscillator modules (this included all individual oscillators).
- p. All telemetry transmitter modules
- q. All tape recorders



#### WBS 4.41.3

The items listed above are by no means a complete list, nor was it intended to present a complete list of troubleshooting methods in this report; rather, it was included to show the many ways the instrumentation engineer had at his disposal to locate and correct a malfunction or discrepancy to the data system or parameter.

#### Inspection

Periodic inspection and periodic maintenance were accomplished on many components of the data system in order to further assure that the system would perform in an accurate and trouble-free manner. The periodic inspection and maintenance of various components were governed by either manufacturers recommendations, or by past performance and history or by a design requirement.

#### Improvements

Improvements in the data acquisition system were a continuous program. All phases of the data system were constantly analyzed to determine if any component or system could be improved to produce a more reliable, accurate, or trouble-free system. The following are examples of improvements that were made to the data system.

- a. Printed circuit card connectors were changed to a more reliable and trouble-free type.
- b. All type transports were modified to permit use of very thin base data tape resulting in greater record time available for in flight data acquisition.
- c. All "CCC" type pressure transducer signal conditioning modules and power supplies were wired for individual fused circuits, to provide protection for each parameter.
- d. Analog data system electronics and tape transport were changed to Ampex type of equipment to provide for greater reliability and interchange of existing equipment used on the digital data system.
- e. A design change was made to the analog to digital converter to improve the digital data accuracy.

#### Flight Test Data Acquisition:

The amount of data acquired in the flight test program proved the desirability of using a high speed system. The entire system recorded an average of 82.5 million data points per flight while an average 22.6 million data points were reduced per flight. Percentage yield of acceptable data averaged 96%. With the aid of these results, aerodynamic, performance, and handling characteristics of the air vehicle were established to high altitude and high Mach numbers. A large amount of information was also obtained on structural integrity, inlet duct and engine performance, and other subsystem operation.



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#### Data Accuracy:

The accuracy of data gathered during the flight test program must be divided into several parts. The digital recording system, excluding sensors, had an input accuracy of  $\pm 0.05\%$  full scale. The recording accuracy of the analog system, excluding sensors, was  $\pm 3\%$  full scale. The sensors had accuracies ranging from 0.001% for the Litton pressure transducers to CCC medium pressure range transducers which were accurate to within 3%. All other data sensors were between these limits.

#### System Reliability:

The entire data acquisition system proved to be very reliable. There was only one instance of system malfunction when the recording tape folded over on one of the digital recorders. Data from this tape were recovered by other than standard data reduction methods. There were records of other malfunctions but, in all cases, the cause was something other than the basic system. In fact, at least two flight plans were changed due to the fail-safe and warning lights built into the system. Incorrect procedures and also failure to follow procedures accounted for the rest of the malfunctions. No flight was completely lost due to a data system malfunction. DRIGINAL PAGE BLACK AND WHITE PHOTOGRAPH



#### COST BREAKDOWNS **B-70 AIRCRAFT STUDY**

#### 4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 03 INSTRUMENTATION

			DESIGN ZENGR HOURS DOLLARS	PROD Hours Dollars	TOTAL Hours Dollars
DESIGN/ENGINEE	RING		84141		84141
LABOR AT \$	4.997		420445		420445
ENGR BURDEN	AT \$	5.458	459205		459205
SHOP SUPPORT			9628		9628
LABOR AT \$	3.444		33160		33160
TEST/QC			417	161	578
LABOR AT \$	3.836		1567	650	2217
MFG BURDEN	AT \$	4.625	46144	1063	47207
ENGR MATERIAL			149948		149948
MPC			24624		24624
SUB-TOTAL			1135093	1713	1136806
GEN & ADMIN			27464	46	27510
TOTAL COST			1162557	1759	1164316

TIME-PHASED COST DETAIL - SEE PAGE II-449 II-457

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING

4-SYSTEM45-SUBSYSTEM416-MAJ ASSY03SUBD OF WORKDESIGN/ENGINEERING

		MAN-	LABOR	LABOR	LABOR	BURDEN	LABOR +
		MONTHS	HOURS	RALE	DULLARS	DULLARS	BURDEN 3
0-3	58	1.5	130	4.292	558	506	1064
Q-4	58						
Q-1	59	1.5	315	4.213	1327	1077	2404
Q-2	59						
Q-3	59	1.5	276	4.236	1169	1006	2175
0-4	59					~	7/00
Q-1	60	6.0	917	4.559	4181	3421	7602
Q-2	60				• • •	1/0	225
Q-3	60		40	4.650	186	149	222
Q-4	60		24	11 700	1100	٥n	1 20 2
Q-1	61		24	40.192	1125	80	1205
0-2	61	7 5	1499	4 000	7101	6497	12588
Q-3	01	(+)	1425	4. 790	1101	0401	10000
0-4	01	22 0	5610	5 000	28566	25777	54343
Q-1	02 40	0.00	5012	2.020	20000	23111	
0-2	6Z 42	33.0	5620	5,181	29119	28464	57583
0-4	62	.J.)•U	5020	J. 101	27123	20101	
0-1	62	43.5	7298	5,256	38356	39544	77900
n = 2	63	1.202	1270				
0-3	63	34.5	5883	5.124	30145	32670	62815
Q-4	63						
0-1	64	36.0	6098	5.214	31797	37190	68 <b>987</b>
0-2	64						
Q-3	64	39.0	6849	5.538	37932	45658	83590
Q-4	64						
Q-1	65	88.5	15302	5.534	84678	100959	185637
Q-2	65						
Q-3	65	88.5	14821	4.907	72721	79773	152494

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING

4-SYSTEM	4	
5-SUB SYSTEM	41	INSTRUMENTATION
6-MAJ ASSY	03	
SUBD OF WORK	DESIGN/ENGINEERI	ING

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-4 6	55						
Q-1 6	56	33.0	5737	3.960	22717	23025	45742
Q-2 6	56						
Q-3 6	56	46.5	7796	3.690	28769	33419	62188
TOTA	AL.	493.5	84141		420445	459205	879650

TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP SUPPORT	
4-SYSTEM	4	
5-SUB SYSTEM	41	INSTRUMENTATION
6-MAJ ASSY	03	
SUBD OF WORK	DESIGN/ENGINEERIN	G

		MAN- MONTHS	LABOR HOUR S	LABOR Rate	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + Burden \$
Q-3 Q-4	64 64		28	3,357	94	389	483
Q-1 Q-2	65 65	37.5	6533	3.494	22828	31661	54489
Q-3 Q-4	65 65	3.0	543	3.363	1826	2341	4167
Q-1 Q-2	66 66	7.5	1380	3.357	4633	9658	14291
Q-3	66	7.5	1144	3.303	3 <b>77</b> 9	2095	5874
тот	AL	55.5	9628		33160	46144	79304

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# TEST/QC 4-SYSTEM 4 5-SUBSYSTEM 41 INSTRUMENTATION 6-MAJ ASSY 03 SUBD OF WORK DESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q-3 64		53	3.208	170		170
Q-1 65	1.5	175	3.560	623		623
Q-2 65 Q-3 65		27	4.519	122		122
Q-4 65 Q-1 66		108	4.324	467		467
Q-2 66 Q-3 66	,	54	3.426	185		185
TOTAL	1.5	417		1567		1567

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM	4	
5-SUB SYSTEM	41 1	INSTRUMENTATION
6-MAJ ASSY	03	
SUBD OF WORK	DESIGN/ENGINEERING	3

		MAN- Mon ths	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$	ENGR Matl
Q-3	58	1.5	130	4.292	558	506	1064	
Q-4	58							
Q-1	59	1.5	315	4.213	1327	1077	2404	
Q-2	59							
Q-3	59	1.5	276	4.236	1169	1006	2175	
0-4	59							
Q-1	60	6.0	917	4.559	4181	3421	7602	
Q-2	60							
Q-3	<b>6</b> 0		40	4.650	186	149	335	
Q-4	<b>6</b> 0	<i>(</i>						
Q-1	61		24	46.792	1123	80	1203	
Q-2	61							
Q-3	61	7.5	1423	4.990	7101	6487	13588	
0-4	61							
Q-1	62	33.0	5612	5.090	28566	25 <b>777</b>	54343	
Q-2	62							
Q-3	62	33.0	5620	5.181	29119	28464	57583	
Q-4	62							
Q-1	63	43.5	7298	5.256	38356	39544	77900	
Q-2	63							
0-3	63	34.5	5883	5,124	30145	32670	62815	
Q - 4	63				~~~	52010	02015	
Q-1	64	36.0	6098	5.214	31797	37190	68987	
0-2	64						00701	
0-3	64	39.0	6930	5,512	38196	46047	84243	54
Q-4	64				00100	10011		24
Q-1	65	127.5	22010	4.913	108129	132620	240749	250.29
0-2	65					**** ** ** **	4.39133	~~~~
$\bar{0} - \bar{3}$	65	91.5	15391	4,851	74669	82114	156783	20080
0-4	65			·• · · *		V LL LL T		20000
Q-1	66	40.5	7225	3.850	27817	32 683	60500	48608

#### **APRIL 1972**

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM45-SUBSYSTEM416-MAJASSY03SUBD0FWORKDESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
Q-2 66 Q-3 66	54.0	8994	3.639	32733	35514	68247	56177
TOTAL	550.5	94186		455172	505 349	960521	149948

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM45-SUBSYSTEM416-MAJ ASSY03SUBD OF WORKDESIGN/ENGINEERING

		MPC	SUB TOTAL	6 8 A	TOTAL
			101/20	0.4.5	
Q-3	58		1064		1064
Q-4	58				
Q-1	59		2404		2404
Q-2	59				
Q-3	59		2175		2175
Q-4	59				
Q-1	60	•	7602	145	7747
Q-2	60				
Q-3	60		335	6	341
Q-4	60				
Q-1	61		1203	22	1225
Q-2	61				
Q-3	61		13588	253	13841
Q-4	61				
Q-1	62		54343	912	55255
0-2	62				
Q-3	62		57583	96.6	58549
Q-4	62				
Q-1	63		77900	1302	79202
Q-2	63				
Q-3	63		62815	1050	63865
Q-4	63				
Q-1	64		68987	1468	70455
Q-2	64				
Q-3	64	20	84317	1794	86111
Q-4	64				
Q <b>-1</b>	65	7486	273264	7291	280555
Q-2	65				
Q-3	65	3582	180445	4814	185259
Q-4	65				
Q-1	66	10081	119189	3590	122779

#### **APRIL 1972**

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# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM45-SUBSYSTEM416-MAJ ASSY03SUBD OF WORKDESIGN/ENGINEERING

		SUB		TOTAL
	MPC	TOTAL	GEA	COST
Q-2 66				
Q-3 66	3455	127879	3851	131730
TOTAL	24624	1135093	27464	1162557

**APRIL 1972** 

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# PRODUCTION

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJASSY 03 SUBD OF WORK PRODUCTION

# INSTRUMENTATION

	MAN-		LABOR		BURDEN	LABOR +
	HUNINS	HUUK S	KALE	DULLARS	DULLARS	BURDEN 3
Q-1 65					811	811
Q-2 65						
Q-3 65					29	29
Q-4 65						
Q-1 66						
Q-2 66						
Q-3 66					223	223
TOTAL					1063	1063

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 4 5-SUBSYSTEM 41 INSTRUMENTATION 6-MAJ ASSY 03 SUBD OF WORK PRODUCTION

ON-SITE LABOR

#### MAN-LABOR LABOR LABOR BURDEN LABOR + MONTHS HOURS RATE DOLLARS DOLLARS BURDEN \$ Q-1 65 121 3.893 471 471 Q-2 65 Q-3 65 6 8.833 53 53 Q-4 65 Q-1 66 1 1 Q-2 66 Q-3 66 34 3.676 125 125 TOTAL 161 650 650

# **APRIL 1972**

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJASSY 03 SUBD OF WORK PRODUCTION

INSTRUMENTATION

		MAN- MON TH S	LABOR HOUR S	LABOR RATE	LABOR Dollars	BUR DEN DOLL ARS	LABOR + BURDEN \$	GEA
Q-1	65		121	3.893	471	811	1282	34
Q-2 Q-3	65 65		6	8.833	53	29	82	2
Q-4 0-1	65 66				1		1	
Q-2	66				-			
Q-3	66		34	3.676	125	223	348	10
тот	<b>AL</b>		161		650	1063	1713	46

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM	4
5-SUBSYSTEM	41
6-MAJ ASSY	03
SUBD OF WORK	PRODUCTION

	INSTRUMENTATION
CTION	

.

	TO TAL COST
Q-1 65	1316
Q-2 65	
Q-3 65	84
Q-4 65	
0-1 66	1
0-2 66	-
Q-3 66	358
TOTAL	1759

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 03 INSTRUMENTATION

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-3	58	1.5	130	4.292	558	506	1064
Q-4	58						
Q-1	59	1.5	315	4.213	1327	1077	2404
Q-2	59	• **					
Q-3	59	1.5	276	4.236	1169	1006	2175
Q-4	59 Ka	4 0	017		4101	2 6 2 1	7/00
0-2	60	0.0	917	4.005	4101	3421	1002
0-3	60		40	4.650	186	149	335
0-4	60		40	4.020	100	147	200
Q-1	61		24	46.792	1123	80	1203
Q-2	61					••••	
Q-3	61	7.5	1423	4.990	7101	6487	13588
Q-4	61						
Q-1	62	33.0	5612	5.090	28566	25777	54343
Q-2	62						
Q-3	62	33.0	5620	5.181	29119	28464	57583
Q-4	62						
Q-1	63	43.5	7298	5.256	38356	39544	77900
Q-2	63						
Q-3	63	34.5	5883	5.124	30145	32670	62815
Q - 4	63	24 0	6000	5 214	21 707	27100	60007
0-1	04 47	20.0	0090	0.214	51 191	57190	00701
0-3	64	39.0	6849	5, 538	37932	45658	83590
0-4	64	57.0	0047	5. 500	2020	12020	0.000
ũ−1	65	88-5	15302	5.534	84678	100959	185637
0-2	65		20000				
Q-3	65	88.5	14821	4.907	72721	79773	152494

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING

4-SYSTEM 4 5-SÜBSYSTEM 41 6-MAJ ASSY 03 INSTRUMENTATION

	MAN- MONTHS	LABOR HOURS	LABUR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-4 65 Q-1 66	33.0	5737	3,960	22717	23025	45749
Q-2 66 Q-3 66	46.5 7796	7796	3.690	28769	33419	62188
TOTAL	493.5	84141		420445	459205	879650

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PRODUCTION 4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 03 IN STRUMENTATION

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1 65		•			811	811
Q-2 65						
Q-3 65					29	29
Q-4 65						
Q-1 66						
Q-2 66						
Q-3 66					223	223
TOTAL					1063	1063

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP	SUPPORT
4-SYSTEM	4	
5-SUB SYSTEM	41	
6-MAJ ASSY	03	
INSTRUMEN	TATION	N

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-3	64		28	3.357	94	389	483
Q <b>-4</b>	64						
Q-1	65	37.5	6533	3.494	22828	31661	54489
Q-2	65						
Q-3	65	3.0	543	3.363	1826	2341	4167
Q-4	65						
0-1	66	7.5	1380	3.357	4633	9658	14291
Q-2	66						
Q-3	65	7.5	1144	3.303	3779	2095	5874
тот	AL	55.5	9628		33160	46144	79304

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 03 INSTRUMENTATION

		MAN- MON TH S	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-3	64		53	3.208	170		170
Q-4	64		<b>26</b> (	<b>n</b>			1.00 (
Q-1	65	1.5	296	3.696	1094		1094
Q-2	65						
Q-3	65		33	5.303	175		175
Q-4	65						
Q-1	66		108	4.333	468		468
Q-2	66						
Q-3	66		88	3.523	310		310
тот	AL	1.5	578		2217		2217

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJASSY 03 INSTRUMENTATION

		MAN- MON TH S	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$	ENGR MATL
Q-3	58	1.5	130	4.292	558	506	1064	
Q-4	58							
0-1	59	1.5	315	4.213	1327	1077	2404	
Q-2	59							
Q-3	59	1.5	276	4.236	1169	1006	2175	
Q-4	59							
Q-1	60	6.0	917	4.555	4181	3421	7602	
Q-2	60							
Q-3	60		40	4.650	186	149	335	. (
Q-4	60							
Q-1	61		24	46.792	1123	80	1203	
Q-2	61							
Q-3	61	7.5	1423	4.990	7101	6487	13588	
Q-4	61							
Q-1	62	33.0	5612	5.090	28566	25777	54343	
Q-2	62							
Q-3	62	33.0	5620	5.181	29119	28464	57583	
Q-4	62	( <b>5 -</b>		_				
	63	43.5	7298	5.256	38356	39544	77900	
	0.3	24 5		<b>-</b> • • •				
0-4	0) 47	54+5	5883	5.124	30145	32670	62815	
0-1	0.) 4 /	26.0	(000	<b>E</b> 01/				
Q-1 0-2	64 67	50.0	0098	5.214	31797	37190	68987	
0-2	64	20.0	(070	5 510	20107			
	64		0420	2+215	38196	46 04 7	84243	54
0-1	65	127 5	22121	1. 007	100/00	100/01	242023	
0-2	65	IC to D	2.4.1.51	4.301	TUSOUC	133431	242031	25029
0-2	65	91.5	15207	1 952	74777	07167	15/0/5	20000
Q-4	65	لہ ہے ہ	12271	4.000	14122	02143	100800	20080
Q-1	66	40.5	7225	3.850	27818	32683	60501	48608

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 03 INSTRUMENTATION

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR Dollars	BUR DEN Doll Ars	LABOR + BURDEN \$	ENGI MATI
Q-2 66 Q-3 66	54.0	9028	3.640	32858	35737	68 <b>5</b> 95	5617
TOTAL	550.5	94347		455822	506412	962234	14994

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 03 INSTRUMENTATION

		MPC	SUB	GEA	TOTAL
			10172	U U A	0051
Q-3	58		1064		1064
Q-4	58				
Q-1	59		2404		2404
Q-2	59				
Q-3	59		2175		2175
Q-4	59				
Q-1	60		7602	145	7747
Q-2	60			,	2/1
Q-3	60		335	6	341
Q-4	60		1 20 2	<b></b>	1 2 2 5
	01 41		1205	22	1225
Q-2	61		12500	252	13841
0-4	61		10000	£.) J	10041
0-1	62		54343	912	55255
$\alpha - 2$	62		21212	/1 L	
0-3	62		57583	966	58549
0-4	62				
Q-1	63		77900	1302	79202
Q-2	63				
Q-3	63		62815	1050	63865
Q-4	63				
Q-1	64		6898 <b>7</b>	1 46 8	70455
Q-2	64				
Q-3	64	20	84317	1794	86111
Q-4	64	74.04	07/5//		201071
Q-1	65	1485	274546	1325	281871
0-2	07	7507	100577	4.01.4	1052/2
Q-3	07	2300	180221	4810	102242
ų−4 ∩−1	65	10081	119190	3590	122780
₩ <b>_1</b>	00	LOOGL	119190	5550	122100

#### APRIL 1972

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 03 INSTRUMENTATION

	MPC	SUB TOTAL	G&A	TOTAL Cost
Q-2 66				
Q-3 66	3455	128227	3861	132088
TOTAL	24624	1136806	27510	1164316

# COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 4
5-SUBSYSTEM 41
6-MAJ ASSY 05
MAJOR AIR VEHICLE REPAIR

		ppnn	TOTAL
	HOURS	HOURS	HOURS
	DELLARS	DOLLARS	DOLLARS
DESIGN/ENGINEERING	9563		9563
LABOR AT \$ 5.472	52325		52325
ENGR BURDEN AT \$ 6.084	58186		58186
SHOP SUPPORT	33500		33500
LABOR AT \$ 3.807	127524		127524
TEST/QC	5158	1154	6312
LABOR AT \$ 4.497	22536	5849.	28385
MFG BURDEN AT \$ 4.757	181775	7620	189395
ENGR MATERIAL	21473		21473
MPC	6093		6093
OTHER COST	60		60
SUB-TOTAL	469972	13469	483441
GEN & ADMIN	12536	359	12895
TOTAL COST	482508	13828	496336

TIME-PHASED COST			
DETAIL - SEE PAGE	II-471	II <b>-</b> 476	II <b>-</b> 480

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

DESIGN/ENGINEERING 4-SYSTEM 4 5-SUBSYSTEM 41 MAJOR AIR VEHICLE REPAIR 6-MAJ ASSY 05 SUBD OF WORK DESIGN/ENGINEERING

#### ON-SITE LABOR

	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + § Burden \$
Q-1 65	43.5	7599	5.816	44198	50920	95118
Q-2 65						
Q-3 65	10.5	1877	4.147	7783	6895	14678
Q-4 65						
Q-1 66		1	. 999	-1	1	
Q-2 66						
Q-3 66		86	4.012	345	370	715
TOTAL	54.0	9563		52325	58186	110511

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### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP SUPPORT
4-SYSTEM	4
5-SUBSYSTEM	41 MAJOR AIR VEHICLE REPAIR
6-MAJ ASSY	05
SUBD OF WORK	DESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-1 65	165.0	28593	3.761	107547	156123	263670
Q-2 65 Q-3 65	28.5	4879	4.077	19894	26061	45955
Q-1 66		24	3.250	78	137	215
Q-3 66		4	1.250	5	- 546	-541
TOTAL	193.5	33500		127524	181775	309 <b>299</b>

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# TEST/QC

4-SYSTEM45-SUBSYSTEM416-MAJ ASSY05SUBD OF WORKDESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1 65	24.0	4163	4.376	18218		18218
Q-2 65 Q-3 65	6.0	1108	4.326	4793		4793
Q-4 65 Q-1 66		7	3.286	23		23
Q-2 66 Q-3 66	5	-120	4.150	-498	·	-498
TOTAL	29.5	5158		22536		22536

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM45-SUBSYSTEM416-MAJ ASSY05SUBD OF WORKDESIGN/ENGINEERING

	MAN- MON THS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BURDEN DOLL ARS	LABOR + Burden \$	ENGR MATL
Q-1 65	232.5	40355	4.212	169963	207043	377006	17666
Q=2 65 Q=3 65	45.0	7864	4.129	32470	<b>329</b> 56	65426	4369
Q = 1 66		32	3.125	100	138	238	443
Q-3 66	5	-30	4.933	-148	-176	-324	-1005
TOTAL	277.0	48221		202385	239961	442346	21473

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 MAJOR AIR VEHICLE REPAIR 6-MAJ ASSY 05 SUBD OF WORK DESIGN/ENGINEERING

	MPC	OTHER COST	SUB TOTAL	G & A	TOTAL COST
Q-1 65	5284	21	399977	10671	410648
Q = 2 65 Q = 3 65	779	19	70593	1883	72476
Q-4 65 Q-1 66	92	20	793	24	817
Q-2 66 Q-3 66	-62		-1391	-42	-1433
TOTAL	6093	60	469972	12536	482 508

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# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# PRODUCTION

4-SYSTEM 4 5-SUBSYSTEM 41 N 6-MAJ ASSY 05 SUBD OF WORK PRODUCTION

MAJOR AIR VEHICLE REPAIR

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	65					7622	7622
Q-4 Q-1	65 66					-1	-1
Q-2 Q-3	65 66					-1	-1
TOT	AL					7620	7620

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# TEST/QC

4-SYSTEM45-SUBSYSTEM416-MAJ ASSY05SUBD OF WORKPRODUCTION

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-3 65	7.5	1154	5.066	5846		5846
Q-4 65						
Q-1 66				1		1
Q-2 66						
Q-3 66				2		2
TOTAL	7.5	1154		5849		5849

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM45-SUBSYSTEM416-MAJASSY05SUBDOFWORKPRODUCTION

	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$	G & A
Q-3 65	7.5	1154	5.066	5846	7622	13468	359
Q = 1 66				1	-1		
Q-3 66				2	-1	1	
TOTAL	7.5	1154		5849	7620	13469	359

#### APRIL 1972

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM	4			
5-SUB SYSTEM	41	MAJOR ATR	VEHTOTE	REPATR
6-MAJ ASSY	05	MADON AIN	A TRUT OF THE	
SUBD OF WORK	PRODUCTION			

TOTAL	
COST	

Q-3 65	13827
Q-4 65	
Q-1 66	
Q-2 66	
Q-3 66	1
TOTAL	13828
NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 05 MAJOR AIR VEHICLE REPAIR

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q = 1 Q = 2	65 65	43.5	7599	5.816	44198	50920	95118
Q-3 Q-4	65 65	10.5	1877	4.147	7783	6895	14678
Q-1 Q-2	66 66		1	• 99 9	- 1	· · · 1	
Q-3	66		86	4.012	345	370	715
тот	AL	54.0	9563		52325	58186	110511

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PRODUCTION 4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 05 MAJOR AIR VEHICLE REPAIR

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	65					7622	7622
Q-4	65 66					-1	-1
Q-2	66						•
Q-3	66					-1	-1
тот	AL					7620	7620

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 05 MAJOR AIR VEHICLE REPAIR

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-1 65 Q-2 65	165.0	28593	3.761	107547	156123	2636 <b>7</b> 0
Q-3 65 0-4 65	28.5	4879	4.077	19894	26061	45955
Q-1 66 Q-2 66		24	3.250	78	137	215
Q-3 66		4	1.250	5	-546	-541
TOTAL	193.5	33500		127524	181775	309299

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 05 MAJOR AIR VEHICLE REPAIR

	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1 65	24.0	4163	4.376	18218		18218
Q-2 65						
Q-3 65	13.5	2262	4.703	10639		10639
Q-4 65						
Q-1 66		7	3.429	24		24
Q-2 66						
Q-3 66	5	-120	4.133	-496		-496
TOTAL	37.0	6312		28385		28385

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 05 MAJOR AIR VEHICLE REPAIR

	MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR Dollars	BUR DEN DOLL AR S	LABOR + BURDEN \$	ENGR MATL
0-1 65	232.5	40355	4.212	169963	207043	377006	17666
Q-2 65 Q-3 65	52.5	9018	4.249	38316	40578	78894	4369
Q-4 65 Q-1 66		32	3.156	101	137	238	443
Q-2 66 Q-3 66	5	-30	4.867	-146	-177	-323	-1005
TOTAL	284.5	49375		208234	247581	455815	21473

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 4 5-SUBSYSTEM 41 6-MAJ ASSY 05 MAJOR AIR VEHICLE REPAIR

	MPC	OTHER COST	SUB TOTAL	GEA	TOTAL COST
Q-1 65 Q-2 65	5284	21	399977	10671	410648
Q-3 65 Q-4 65	779 -	19	84061	2242	86303
Q = 1 66 Q = 2 66	92	20	793	24	817
Q-3 66	-62		-1390	-42	-1432
TOTAL	6093	. 60	483441	12895	496336





#### WBS 5.0: FLIGHT TEST GROUND SUPPORT EQUIPMENT

#### OBJECTIVES AND SCOPE:

Flight Test Ground Support Equipment (FTGSE) included all test equipment items provisioned for the support of the XB-70 air vehicles during the flight test development program. The FTGSE was provisioned based on the following objectives:

- (1) To provide FTGSE for support of the air vehicle and installed subsystems during the flight test program.
- (2) To provide FTGSE for maintenance of the components determined to be maintainable within the NR, Los Angeles Division complex.

The ground rules which were applied in developing equipment to meet the objectives were that all FTGSE necessary to support on-site maintenance operations would be provisioned; however, to accomplish this objective within the severe funding restrictions, maximum use was to be made of GFP, NR technician skills, and the sharing of manufacturing tools (TSC) between Manufacturing and Engineering Flight Test. The approach was to confine all risks involved to possible flight schedule impacts with no compromise to maintenance of the air vehicle as related to flight safety.

The XB-70 FTGSE program evolved when in December, 1959, the B-70 Program was redirected from a production status to a flight test development status which automatically deleted all requirements for manufacturing type Ground Support Equipment (GSE). With the redirection, funding for the XB-70 program was critical which dictated that the capability be provided for maximum utilization (interdepartment transfer or sharing) of manufacturing tooling and special test equipment. In addition, to further reduce cost, if analysis showed it more economical to originally configure a special test equipment item for flight test use, it was provisioned as FTGSE and then utilized by Manufacturing for system installation and/cr checkout.

The procurement or provisioning of FTGSE, as directed by the XB-70 contract, was based on "lead-time-away" from the required date of the individual end item. Under this ground rule, the objective of the procurement program for the testing equipment was to have each item of FTGSE on site 30 days prior to first use during either final checkout and preflight or flight test operations. (The 30 days prior delivery was utilized for checkout, calibration, and Air Force Inspection approval). The procurement of FTGSE was categorized by the major subsystems which they supported, such as, air vehicle handling, alighting and arresting, flight control, central air data, propulsion, secondary power, environmental control, air induction control, mission and traffic control, and personnel protection. The FTGSE provisioned was of two types: that which directly supported the air vehicle installed subsystems, and that for subsystem component repair. The following table shows the FTGSE breakdown as related to CFE and GFP items at the time period of air vehicle No. 1 preflight.

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Source		End Items
CFE Air Vehicle FTGSE CFE Component Repair FTGSE GFP FTGSE		652 145 299
	TOTAL	1096

The 1096 end items provisioned at this "out-the-door" point in time increased to a total of 1600 end items during the course of the flight test program. The increase was essentially due to: (1), knowledge gained in subsystem operation showing additional requirements; and (2), the closing out of subcontractors and assuming "in-house" responsibility for the component repair. Of the total 1600 end items of FTGSE, approximately 1000 would be classified as 'special' while the other 600 end items would be classified as conventional type GSE.

The equipment required for test, checkout, and servicing of the air vehicle is summarized in the following paragraphs.

Air Vehicle Handling Equipment: This equipment included FTGSE such as that required for air vehicle towing, weighing, leveling, mooring, and jacking, as well as that necessary for gaining access to the various equipment service connections and for protective covering.

Personnel Protection: Equipment in this category included that necessary for servicing the pressurization, gaseous oxygen, and liquid oxygen systems, accomplishing capsule pressurization system checkout and electrical systems checkout.

Alighting and Arresting: FTGSE in this category included all equipment necessary for servicing and checkout of all systems associated with the landing gear including nose wheel steering, braking system, hot-tire sensing system, struts, and tire servicing. It also included that equipment required for drag chute installation and for drag chute deployment.

Secondary Power: There were several categories of equipment required for service and checkout of the secondary power system as well as for continuous support of the air vehicle during maintenance of the other subsystems in the hangar and at remote sites. The FTGSE included equipment for constant speed drive loading, hydraulic system leakage and functional checks, hydraulic pump test set, hydraulic fluid servicing, filter cart, and gaseous nitrogen servicing trailers. Electrical system checkout included ground power unit, phase rotation tester, electrical power test set, and power analyzer and test set. In addition, several types of equipment were required for air vehicle wiring and power distribution checkout and/or de-bugging.

Propulsion: The FTGSE required for propulsion subsystem servicing, checkout, and test included engine lubrication and hydraulic servicing carts, sound supressors, FOD screens, fuel deaeration, gaseous nitrogen servicing, fuel

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sampling, ullage gas analyzing, liquid nitrogen servicing, low pressure tester, fuel filtration, various flow meters and the GE "suitcase".

Flight Control: The flight control test sets included equipment for checking such items as surface travel, rate, force and deflection as well as functional testing of the FACS, ancillary controls, and the trim systems.

Mission and Traffic Control: The equipment was essentially all GFP and was required for pre-flighting the Intercom, UHF, ILS, TACAN, and IFF systems.

Central Air Data: The equipment required for the CADS was for testing the pressure sensors, total temperature probe, computer, indicators, and amplifiers.

Air Induction System: FTGSE for the AIS included that equipment required to checkout, test, and calibrate the AICS sensors, probes, surface travels and rates for both the automatic control, standby and emergency systems.

Environmental Control: The diverse requirements for ECS FTGSE fell into two categories: that required for system checkout on the air vehicle and that required for engineering evaluation. This equipment included such items as low pressure testers for leakage, air flow calibraters, cooling effect detectors testers, air flow distribution analyzer, laminar flow meters, motor-generator set, ammonia and heated water servicing, gas turbine and hose sets, freon system tester and several adapters.

The FTGSE for electrical and mechanical component repair was comprised of those end items for: (1), isolation and removal of the faulty component; (2), repair of the component by replacement of either assemblies or bits and pieces; (3), component reverification to confirm repair; and (4), air vehicle installation and subsystem performance verification.

Exhibits 1 and 2, pages II-490 and II-491, respectively present typical FTGSE arrangements for mission start and mission completion. Exhibits 3 through 12 present pictures of some typical FTGSE.

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FTGSE - MISSION START



# II-490

# EXHIBIT 1

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EXHIBIT 2



# ORIGINAL PAGE BLACK AND WHITE PHOTOGRAPH















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# ORIGINAL PAGE BLACK AND WHITE PHOTOGRAPH





# COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 5 GROUND SUPPORT EQUIPMENT

			5-SUBSYS	5-SUBSYS	
			50	51	T OT AL
			HOURS	HOURS	HOURS
			DOLLARS	DOLLARS	DOLLARS
DESIGN/ENGINEER	RING		1580700	83812	664512
LABCR AT \$	4.407		6894024	440678	7334702
ENGR BURDEN	AT \$	4.852	7561259	515432	8076691
PRODUCTION			224559		224559
LABOR AT \$	3.326		746798		746798
SHOP SUPPORT			272322	21443	293765
LABOR AT \$	4.706		1287067	954.29	1382496
PLANN ING			7858		7858
LABOR AT \$	3.633		28551		28551
TEST/QC			14866	4891	19757
LABOR AT \$	3.572		52560	18007	70567
MFG BURDEN	AT \$	4.471	2313564	127366	2440930
ENGR MATERIAL			216590	7532	224122
MFG MATERIAL			1578449		1578449
MPC			233560	1866	235426
OTHER COST			60604		60604
SUB-TOTAL			20973026	1206310	22179336
GEN & ADMIN			397683	26637	424320
TOTAL COST			21370709	1232947	22603656

SUBDIVISION OF WORK

COST DETAIL - SEE PAGE II-503 II-537 II-547

# COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 5 5-SUBSYSTEM 50 GSE BASIC

			DESIGN /ENGR HOURS DOLLARS	PROD Hours Dollars	TEST /QC HOURS DOLLARS	TOTAL Hours Dollars
DESIGN/ENGINEE	RING 4-361		1576734 6876930	3854 16586	112	L580700 6894024
ENGR BURDEN	AT \$	4.783	7540100	20734	425	7561259
PRODUCTION				224559		224559
LABOR AT \$	3.326			746798		746798
SHOP SUPPORT			271960		362	272322
LABOR AT \$	4.726		1286027		1040	1287067
PLANN ING			,	7858		<b>78</b> 58
LABOR AT \$	3.633			28551		28551
TEST/QC			714	14138	14	14866
LABOR AT \$	3.536		2710	49785	65	52560
MFG BURDEN	AT \$	4.453	1344265	967922	1377	2313564
ENGR MATERIAL			214558		2032	216590
MFG MATERIAL				1578449		1578449
MPC			28502	204948	110	233560
OTHER COST			60550	54		60604
SUB-TOTAL			17353692	3613827	5507	20973026
GEN & ADMIN			327086	70575	22	397683
TOTAL COST			17680778	3684402	5529	21370709

TIME-PHASED COST				
DETAIL-SEE PAGE	II-504	II-514	II <b>-</b> 520	II <b>-</b> 525

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESIGN/ENGINEERING			
4-SYSTEM	5			
5-SUB SYSTEM	50	GSE	BASIC	
SUBD OF WORK	DESIGN/ENGINEERING			

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + Burden \$
Q-1	53	37.5	6362	4.366	27779	28944	56723
Q-2	58						
0-3	58	268.5	45069	4.153	187177	178709	365886
Q-4	58						
Q-1	59	480.0	81982	4.041	331270	281781	613051
Q-2	59						
Q-3	59	766.0	134772	3.990	537695	482324	1020019
0-4	59						
Q-1	60	439.5	76161	4.500	342727	285751	628478
Q-2	60						
Q-3	60	261.0	43754	5.127	224326	162162	386488
Q-4	60						
Q-1	61	711.0	121378	4.612	559767	410982	970749
Q-2	61						
Q-3	61	465.0	84321	4.946	417049	394373	811422
Q-4	61						
Q-1	62	568.5	97112	4.889	474778	445240	920018
Q-2	62						
Q-3	62	82.5	13826	3.700	51159	82978	134137
Q-4	62						
0-1	63	126.0	21483	3.967	85223	109934	195157
Q-2	63						
Q-3	63	349.5	58740	4.955	291075	350905	641980
Q-4	63						
Q-1	64	478.5	81608	5.312	433476	488455	921931
Q-2	64						
Q-3	64	304.5	53502	5.738	306982	354976	661958
Q-4	64						
Q-1	65	3498.0	606198	3.928	2381016	3145245	5526261
Q-2	65						

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

DESIGN/ENGINEERING 4-SYSTEM 5 5-SUBSYSTEM 50 GSE BASIC SUBD OF WORK DESIGN/ENGINEERING

LABOR + BURDEN \$	BUR DEN DOLL ARS	LA80R DOLLARS	LABOR RATE	LABUR HOURS	MAN- MONTHS	
437775	257438	180337	5.004	36042	214.5	Q-3 65
73848	44265	29583	2.797	10578	61.5	Q-4 65 Q-1 66
51199	35638	15561	4.046	3846	22.5	Q-2 66 Q-3 66
14417080	7540100	6876980		1576734	9134.5	TOTAL

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP SUPPORT	
4-SYSTEM	5 (	SE BASIC
5-SUB SYSTEM	50	
SUBD OF WORK	DESIGN/ENGINEERING	3

		MAN- MON TH S	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-1	58		4	2.000	8	14	22
0-2	58						
Q-3	58		36	3.278	118	146	264
Q-4	58						
Q-1	59	130.5	2215 <b>7</b>	2.725	60370	81977	142347
Q-2	59						
Q-3	59	82.5	14425	2.748	39643	61617	101260
Q-4	59						
Q-1	60	9.0	1650	2.855	4710	6545	11255
Q-2	60						
Q-3	60					-386	-386
Q-4	60						
Q-1	61						
Q-2	61						
Q-3	61		11	2.455	27	42	69
Q-4	61						
Q-1	62	42.0	7117	5.006	35625	32881	68506
Q-2	62						
Q-3	62	546.0	91772	4.996	458493	447996	906489
Q-4	62						
Q-1	63	501.0	85585	5.237	448246	460 187	909033
Q-2	63				222/22	200544	( ) 0 ) 0 (
Q-3	63	241.5	40571	5.167	209630	208564	418194
Q-4	63			~ • • • •		7005	11707
Q-1	64	9.0	1444	3.111	4492	1235	11/2/
Q-2	64		( 0 0	2 27/	3334	3 <b>3 7</b> 7	5701
Q-3	64	4.5	689	3.316	2326	3315	1010
Q-4	64	21 0	34.00	7 (71	15/01	10010	21600
<u>v-1</u>	65	21.0	3699	3.431	12091	19218	21004
Q = 2	65						

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 5 GSE BASIC 5-SUBSYSTEM 50 SUBD OF WORK DESIGN/ENGINEERING

	MAN- MON TH S	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 65 Q-4 65	9.0	1425	3.286	4683	6909	11592
Q = 1 66 Q = 2 66	1.5	277	4.170	1155	1649	2804
Q-3 66	6.0	1098	3.470	3810	5996	9806
TOTAL	1603.5	271960		1286027	1344265	26302 <b>92</b>

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 5 GSE BASIC 5-SUBSYSTEM 50 SUBD OF WORK DESIGN/ENGINEERING

		MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	58		4	2.250	9		9
0-4	58						
Q-1	59						
Q-2	59						
Q-3	59		26	2.846	74		74
Q-4	59						
Q-1	60		9	3.889	35		35
Q-2	60						
Q-3	60						
Q-4	60						
Q-1	61						
Q-2	61						
Q-3	61						
Q-4	61						
Q-1	62			ý Ř			
Q-2	62			• · · . - 			
Q-3	62			- 7. - 7. - 7.			
Q-4	62						
Q-1	63		47	4.681	220		220
Q-2	63						
Q-3	63		26	5.462	142		142
Q-4	63				· · · · ·		~~~~
Q-1	64		10	3.243	227		227
Q-2	64			<b>D 1 D</b> (	~ <b>~</b>		00
Q-3	64		31	3.194	99		99
Q-4	54			<b>a</b> 0 <b>a</b> 0	1100		1100
Q-1	00	1.0	311	3 • 83 3	1195		1192
Q-2	0) (5		0.2	2 425	214		214
Q-3	07		92	2.435	21.0		210
<b>U</b> -4	07						

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# TEST/QC 4-SYSTEM 5 GSE BASIC 5-SUBSYSTEM 50 SUBD OF WORK DESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
0-1 66		29	4.345	126		126
Q-2 66 Q-3 66		69	3.913	270		270
TOTAL	1.5	714		2710		2710

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM5GSE BASIC5-SUBSYSTEM50SUBD OF WORKDESIGN/ENGINEERING

		MAN- MON TH S	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR Matl
Q-1	58	37.5	6366	4,365	27787	28958	56745	
Q-2	58							
0-3	58	268.5	45109	4.152	187304	178855	366159	
Q-4	58							
Q <b>-1</b>	59	610.5	104139	3.761	391640	363758	755398	14981
Q-2	59							
Q-3	59	848.5	149223	3.869	577412	543941	1121353	5254
Q-4	59							
Q-1	60	448.5	77820	4.465	347472	292296	639768	1144
Q-2	<b>6</b> 0							
Q-3	60	261.0	43754	5.127	224326	161776	386102	-1
Q-4	60							(
Q-1	61	711.0	121378	4.612	559767	410982	970749	-12
Q-2	61							
Q-3	61	465.0	84332	4.946	417076	394415	811491	441
Q-4	61							
0-1	62	610.5	104229	4.897	510403	478121	988524	-4
Q-2	62							
Q-3	62	628.5	105598	4.826	509652	530974	1040626	975
Q-4	62							
Q-1	63	627.0	107115	4.982	533689	570721	1104410	53198
Q-2	63							
Q-3	63	591.0	99337	5.042	500847	559469	1060316	42977
Q-4	63							
Q-1	64	487.5	83122	5+272	438195	495690	933885	29476
Q-2	64	~~~						
Q-3	64	309.0	54222	5.706	309407	358351	667758	/130
Q-4	64	2520 5	(10000					
1-V	65	3520.5	610208	3.925	2394899	3164163	5559062	9121
Q-2	65	<b>00</b> 0 F				0// 0/ T		
Q-3	65 (5	223.5	31229	4.935	185336	264347	449683	45197
Q-4	65							

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM5GSE BASIC5-SUBSYSTEM50SUBDOF WORKDESIGN/ENGINEERING

ENGR MATL	LABOR + BURDEN \$	BUR DEN DOLL ARS	LABOR DCLLARS	LABOR RATE	LABOR HOUR S	MAN- Months	
1150	76778	45914	30864	2.836	10884	63.0	Q-1 66
2931	61275	41634	19641	3.918	5013	28.5	Q-2 66 Q-3 66
214558	17050082	8884365	8165717		1849408	10739.5	TOTAL

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 5 GSE BASIC 5-SUBSYSTEM 50 SUBD DF WORK DESIGN/ENGINEERING

		мрс	OTHER COST	SUB Total	G & A	TOTAL COST
Q-1	58			56745		56 <b>7</b> 45
Q-2	58					
Q-3	58			366159		366159
Q-4	58					
Q-1	59	1269		771648		771648
Q-2	59					
Q-3	59	445	4629	1131681		1131681
Q-4	59					
Q-1	60	150	9366	650428	12393	662821
Q-2	60					
Q-3	60		3025	389126	7414	396540
Q-4	60	_				
Q-1	61	-1	7379	978115	18176	995291
0-2	61					
Q-3	61	37	4277	816246	15168	831414
Q-4	61					
Q - 1	62		2608	991128	16636	1007764
Q-2	62				ست من من من م	
Q-3	62	/6	7757	1049434	17615	1067049
Q-4	62					
Q-1	63	5240	5638	1168486	19520	1188006
Q-2	63				* * * * *	
Q-3	63	4233	7851	1115377	18649	1134026
Q-4	63					
Q-1	64	3142	4616	971119	20663	991782
Q-2	64					
Q = 3	64	2594	3108	680590	14482	695072
Q-4	64					
Q-1	65	2728	17	5570928	148632	5719560
Q-2	65		<u> </u>		1.5.1.4	أنعر ومعروب والم
Q-3	65	8170	157	503807	13442	517249
Q-4	65					

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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**TIME PHASED EXPEND.** B-70 AIRCRAFT STUDY

4-SYSTEM5GSE BASIC5-SUBSYSTEM50SUBD OF WORKDESIGN/ENGINEERING

		OTHER	SUB		TOTAL
	MPC	COST	TOTAL	G & A	COST
Q-1 66	239	43	78210	2355	80565
Q-3 66	180	79	64465	1941	66406
TOTAL	28502	60550	17353692	327086	17680778

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## DESIGN/ENGINEERING 4-SYSTEM 5 5-SUBSYSTEM 50 GSE BASIC SUBD OF WORK PRODUCTION

#### ON-SITE LABOR

		MAN- MON TH S	LABOR HOURS	LABOR RATE			LABOR +
Q-1	62		8	3.000	24	34	58
Q-2	62						,
Q-3	62	1.5	138	3.145	434	584	1018
Q-4	62						
Q-1	63		-36	3.333	-120	-21	-141
Q-2	63						
Q-3	63	9.0	1526	4.290	654 <b>7</b>	7108	13655
Q-4	63						
Q-1	64	6.0	1138	4.651	5293	6730	12023
Q-2	64						
Q-3	64	3.0	401	3.736	1498	2257	3755
Q-4	64				- · · -		
Q-1	65	1.5	275	4.382	1205	1830	3035
Q-2	65						0000
Q-3	65	1.5	168	4.899	823	1.085	1908
Q-4	65						
Q - 1	66		113	3.920	443	644	1087
Q-2	66						2001
Q-3	66		123	3.569	.439	483	922
тот	AL	22.5	3854		16586	20734	37320

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	PRODUCTION		
4–SYSTEM 5–SUBSYSTEM	5 50	GSE BASIC	
SUBD OF WORK	PRODUCTION		

		MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-3	61	1.5	240	7.238	1737	295	2032
Q-4	61						
Q-1	62	40.0	6783	3.052	20699	24301	45000
Q-2	62						
Q-3	62	309.0	52028	3.027	157511	204229	361740
Q-4	62						
Q-1	63	342.0	58394	3.134	183018	231964	414982
Q-2	63						
Q-3	63	188.5	31620	3.315	104820	140106	244926
Q-4	63						
Q-1	64	189.0	32299	3.573	115404	139808	255212
Q-2	64						
Q-3	64	124.5	21951	3.717	81602	94090	175692
Q-4	64						
Q-1	65	36.0	6363	4.182	26613	27484	5409 <b>7</b>
Q-2	65						
Q-3	65	39.0	6604	3.421	22590	34661	57251
Q-4	65						
Q-1	66	31.5	5502	4.199	23102	32705	55807
Q-2	66						
Q-3	66	16.5	2775	3.496	9702	14054	23 <b>756</b>
TOTAL		1317.5	224559		746798	943697	1690495
# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PLANNING 4-SYSTEM 5 5-SUBSYSTEM 50 SUBD OF WORK PRODUCTION

# GSE BASIC

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1 52	4.5	751	3.688	2770	330	3100
Q-2 62						
Q-3 62	7.5	1272	3.106	3951	1073	5024
Q-4 62						
Q-1 63						
Q-2 63						
Q-3 63	15.0	2616	4.043	10576	7 3 9 1	17967
Q-4 63						
Q-1 64	7.5	1156	3.548	4102	5478	9580
Q-2 64						
Q-3 64	3.0	528	3.371	1780	2432	4212
0-4 64						
Q-1 65	1.5	386	3.443	1329	1821	3150
Q-2 65						
Q-3 65	3.0	470	3.311	1556	2174	3730
Q-4 65						
Q-1 66	3.0	449	3.666	1.646	2427	4073
Q-2 66						
Q-3 66	1.5	230	3.657	841	1099	1940
TOTAL	46.5	7858		28551	24225	52776

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 5 GSE BASIC 5-SUBSYSTEM 50 SUBD OF WORK PRODUCTION

		MAN- MONTHS		LABOR			LABOR +
				6741E	DOLLANG	DOLLAND	
Q-1	62	1.5	273	3.209	876		876
Q-2	62						
Q-3	62	12.0	1976	3.121	6167		6167
Q-4	62						
Q-1	63	12.0	2108	3.759	7923		7923
Q-2	63						
Q-3	63	7.5	1318	2.666	3514		3514
Q-4	63						
Q-1	64	18.0	2995	3.475	10409		10409
Q-2	64						
Q-3	64	12.0	2147	3.499	7513		. 7513
Q-4	64		•				
Q-1	65	4.5	795	3.523	2801		2801
Q-2	65						
Q-3	65	6.0	979	3.769	3690		3690
0-4	65						
Q-1	66	4.5	878	4.257	3738		3738
Q-2	66						
Q-3	66	4.5	669	4.714	3154		3154
тот	AL	82.5	14138		49785		49785

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

GSE BASIC

4-SYSTEM 5 5-SUBSYSTEM 50 SUBD OF WORK PRODUCTION

		MAN- MON TH S	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	MFG Matl
Q-3 Q-4	61 61	1.5	240	7.238	1737	295	2032	
Q-1 Q-2	62 62	46.0	7815	3.118	24369	24665	49034	10523
Q-3 Q-4	62 62	330.0	55414	3.033	168063	205886	373949	166760
Q-1 Q-2	63 63	354.0	60466	3.156	190821	231943	422764	328048
Q-3 Q-4	63 63	220.0	37080	3.383	125457	154605	2800 <b>62</b>	446157
Q-1 Q-2	64 64	220.5	3 <b>7</b> 58 <b>8</b>	3.597	135208	152016	287224	275255
Q-3 Q-4	64 64	142.5	2502 <b>7</b>	3.692	92393	98779	191172	110990
Q-1 Q-2	65 65	43.5	7819	4.086	31948	31 135	63083	74758
Q-3 Q-4	65 65	49.5	8221	3.486	28659	37920	66579	49653
0-1 Q-2	66 66	39.0	6942	4.167	28929	35 <b>776</b>	64705	45444
Q-3	66	22.5	3797	3.723	14136	15636	29772	70861
TOT	AL	1469.0	250409		841720	988656	1830376	1578449

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

> > GSE BASIC

4-SYSTEM 5 5-SUBSYSTEM 50 SUBD OF WORK PRODUCTION

OTHER SUB TOTAL MPC COST TOTAL G & A COST Q-3 61 2032 38 2070 Q-4 61 Q-1 62 829 455 1021 60841 61862 Q-2 62 Q-3 62 13140 -70 553779 9295 563074 Q-4 62 Q-1 63 32312 779 783903 13107 797010 Q-2 63 Q-3 63 43946 -1136 769029 12858 781887 Q-4 63 Q-1 64 29342 16 591837 12593 604430 Q - 2 64Q-3 64 40378 6 342 54 6 7289 349835 Q-4 64 Q-1 65 22360 4 160205 4274 164479 Q-2 65 Q-3 65 8858 125090 3337 128427 Q-4 65 Q-1 66 9425 119574 3601 123175 Q-2 66 Q-3 66 4358 104991 3162 108153 TOTAL 204948 54 3613827 70575 3684402

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESIGN/ENGINEERING		
4-SYS TEM	5		· .
5-SUB SYSTEM	50	GSE	BASIC
SUBD OF WORK	TEST/QC		

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-3	60 60		75	4.547	341	280	621
Q-1	61		37	3.162	117	145	262
тот	AL		112		458	425	883

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

GSE BASIC

SHOP SUPPORT 4-SYSTEM 5 5-SUBSYSTEM 50 SUBD OF WORK TEST/QC

		MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 Q-4	58 58		24	2.208	53	88	141
Q-1 Q-2	59 59	1.5	188	3.037	571	695	1266
Q-3 Q-4	59 59		105	2.438	256	501	757
Q-1 Q-2	60 60		37	3.622	134	66	200
Q-3 Q-4	60 60					-3	-3
Q-1	61		8	3.250	26	30	56
тот	AL	1.5	362		1040	1377	2417

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 5 5-SUBSYSTEM 50 SUBD DF WORK TEST/QC

# GSE BASIC

	MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 59		14	4.643	65		65
TOTAL		14		65		65

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 5 5-SUBSYSTEM 50 SUBD OF WORK TEST/QC

GSE BASIC

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
Q-3 58 Q-4 58		24	2.208	53	88	141	5
Q-1 59 Q-2 59	1.5	188	3.037	571	695	1266	2016
Q-3 59 Q-4 59		119	2.697	321	501	822	-1
Q = 1 60 Q = 2 60		37	3.622	134	66	200	6
Q-3 60 Q-4 60		75	4.547	341	277	618	6
Q-1 61		45	3.178	143	175	318	
TOTAL	1.5	488		1563	1802	3365	2032

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

GSE BASIC

4-SYSTEM 5 5-SUBSYSTEM 50 SUBD OF WORK TEST/QC

	MPC	SUB TOTAL	GεA	TOTAL COST
Q-3 58		146		146
Q-4 58 Q-1 59	110	3392		3392
Q-2 59				
Q-3 59 Q-4 59		821		821
Q-1 60		206	4	210
Q-3 60		624	12	636
Q-4 60 Q-1 61		318	6	324
TOTAL	110	EE 07	2.2	~~~~
TUTAL	110	2201	22	2265

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# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING

4-SYSTEM 5 5-SUBSYSTEM 50 GSE BASIC

		MAN-	LABOR	LABOR	LABOR	<b>BUR DEN</b>	LABOR +
		MONTHS	HOURS	RATE	DOLLARS	DOLL ARS	BURDEN \$
Q-1	58	37.5	6362	4.366	27779	28944	56723
Q-2	58						
Q-3	58	268.5	45069	4.153	187177	178709	365886
Q-4	58						
Q-1	59	480.0	81982	4.041	331270	281781	613051
Q-2	59						
Q-3	59	766.0	134772	3.990	537695	482324	1020019
Q-4	59						
Q-1	60	439.5	76161	4.500	342727	285751	628478
Q-2	60						
Q-3	60	261.0	43829	5.126	224667	162442	387109
Q-4	60		• •				
Q-1	61	711.0	121415	4.611	559884	411127	971011
Q-2	61	4					
Q-3	61	465.0	84321	4.946	417049	394373	811422
Q-4	61						
Q - 1	62	568.5	97120	4.889	474802	445274	920076
Q-2	62						
Q-3	62	82.5	13964	3.695	51593	83562	135155
Q-4	62						
Q-1	63	126.0	21447	3.968	85103	109913	195016
Q-2	63			,			
Q-3	63	358.5	60266	4.938	297622	358013	655635
Q-4	63						
Q-1	64	484.5	82746	5,303	438769	495185	933954
Q-2	64						
Q-3	64	306.0	53903	5.723	308480	357233	665713
Q-4	64						
Q-1	65	3499.5	606473	3.928	2382221	3147075	5529296
$\Omega - 2$	65						

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESIGN/ENGINEERING
4-SYSTEM	5
5-SUB SYSTEM	50
GSE BASIC	

	MAN- MON THS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 65 Q-4 65	216.0	36210	5.003	181160	258523	439683
Q-1 66	61.5	10691	2.809	30026	44909	74935
Q-3 66	24.0	3969	4.031	16000	36121	52121
TOTAL	9155.5	1580700		6894024	7561259	14455283

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	PRODUCTION
4-SYSTEM	5
5-SUB SYSTEM	50
GSE BASIC	

		MAN- MONTHS	LABOR HOURS	LABOR	LABOR	BUR DEN DOLL ARS	LABOR + BURDEN \$
0-3	61	1.5	240	7,238	1737	295	2032
0 - 4	61		2,0	10230	1137	277	2052
0-1	62	40.0	6783	3,052	20699	24 30 1	45000
0-2	62	1000	0105	3.022	20097	21201	12000
0-3	62	309.0	52028	3.027	157511	204229	361740
0-4	62	50710	2020	30021		And the distance of	302110
Q-1	63	342.0	58394	3,134	183018	231964	414982
Q-2	63		2-74-74				
0 - 3	63	188.5	31620	3.315	104820	140106	244926
0-4	63						
Q-1	64	189.0	32299	3.573	115404	139808	255212
Q-2	64						
Q-3	64	124.5	21951	3.717	81602	94090	175692
Q-4	64						
Q-1	65	36.0	6363	4.182	26613	27 48 4	54097
Q-2	65						
Q-3	65	39.0	6604	3.421	22590	34661	57251
Q-4	65						
Q-1	66	31.5	5502	4.199	23102	32705	55 80 <b>7</b>
Q-2	66						
Q-3	66	16.5	2775	3.496	9702	14054	23756
тот	AL	1317.5	224559	·	746798	943697	1690495

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 5 5-SUBSYSTEM 50 GSE BASIC

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOP DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58		4	2.000	8	14	22
Q-2	58						
Q-3	58		60	2.850	171	234	405
Q-4	58						
Q-1	59	130.5	22345	2.727	60941	82672	143613
Q-2	59						
Q-3	59	82.5	14530	2.746	39899	62118	102017
Q-4	59						
Q-1	60	10.0	1687	2.871	4844	6611	11455
Q-2	60						
Q-3	60					-389	-389
Q-4	60						
Q-1	61		8	3.250	26	30	56
Q-2	61						
Q-3	61		11	2.455	27	42	69
Q-4	61						
Q-1	62	42.0	7117	5.006	35625	32881	68506
Q-2	62						
Q-3	62	546.C	91772	4.996	458493	447996	906489
Q-4	62						
Q-1	63	501.0	85585	5.237	448246	460787	909033
Q-2	63						
Q-3	63	241.5	40571	5.167	209630	208564	418194
Q-4	63						
Q-1	64	9.0	1444	3.111	4492	7235	11727
Q-2	64						
Q-3	64	4.5	689	3.376	2326	3375	5701
Q-4	64						
Q-1	65	21.0	3699	3.431	12691	18918	31609
Q-2	65				×.		

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP	SUPPORT
4-SYSTEM	5	
5-SUB SYSTEM	50	
GSE BASIC		

#### ON-SITE LABOR

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 0-4	65 65	9.0	1425	3.286	4683	6909	11592
Q-1 Q-2	66 66	1.5	277	4.170	1155	1649	2804
Q-3	66	6.0	1098	3.470	3810	5996	9806
тот	AL	1604.5	272322	•	1287067	1345642	2632709

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

PLANNING 4-SYSTEM 5 5-SUBSYSTEM 50 GSE BASIC

#### ON-SITE LABOR

	Λ.	MAN- MONTHS	LABOR HDURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	62	4.5	751	3.688	2770	330	3100
Q-2	62						
Q-3	62	7.5	1272	3.106	3951	1073	5024
Q-4	62						•
Q-1	63						
Q-2	63						
Q-3	63	15.0	2616	4.043	10576	7 3 9 1	17967
0-4	63						
0-1	64	7.5	1156	3.548	4102	5478	9580
Q-2	64						
Q-3	64	3.0	528	3.371	1780	2432	4212
Q-4	64						
Q-1	65	1.5	386	3.443	1329	1821	3150
Q-2	65						
Q-3	65	3.0	470	3.311	1556	2174	3730
0-4	65						
Q-1	66	3.0	449	3.666	1646	2427	4073
Q-2	<b>6</b> 6						
Q-3	66	1.5	230	3.657	841	1099	1940
тот	ΓAL	46.5	7858		28551	24225	52 <b>77</b> 6

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 5 5-SUBSYSTEM 50 GSE BASIC

		MAN-				BURDEN	LABOR +
		MONTHS	HUUK 3	RAIL	DULLARS	DOLLARS	BONDEN \$
Q-3	58		4	2.250	9		9
0-4	58						
0-1	59						
Q-2	59						
Q-3	59		40	3.475	139		139
Q-4	59						
Q-1	60		9	3.889	35		35
Q-2	60						
Q-3	60						
Q-4	60						
Q-1	61						
Q-2	61						
Q-3	61						
Q-4	61				~		
Q-1	62	1.5	273	3.209	876		876
0-2	62						
Q-3	62	12.0	1976	3.121	6167		6167
Q-4	62						
Q-1	63	12.0	2155	3.119	8143		8143
Q-2	63				2151		2/5/
Q-3	63	1.5	1344	2.120	3020		2020
Q-4	63	10.0	2015	2 170	10626		10626
	04	18.0	5005	2.410	10020		10000
0-2	0 <del>4</del> 47	12.0	2179	2 405	7612		7612
	04 64	12.0	2110	20772	1012		
0-1	65	6.0	1106	3,610	3993		3993
0-2	65		1100	34010			
0-3	65	6.0	1071	3.740	4006		4006
<u> </u>	65						

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 5 5-SUBSYSTEM 50 GSE BASIC

	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-1 66 Q-2 66	4.5	907	4.260	3864		3864
Q-3 66	4.5	738	4.640	3424		3424
TOTAL	84.0	14866		52560		52560

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### 4-SYSTEM 5 5-SUBSYSTEM 50 GSE BASIC

		MAN-	LABOR	LABOR	LABOR	<b>BUR DEN</b>	LABOR +	ENGR
		MONTHS	HOURS	RATE	DOLLARS	DOLL ARS	BURDEN \$	MATL
Q-1	58	37.5	6366	4.365	27787	28958	56745	
Q-2	58							
Q-3	58	268.5	45133	4.151	187357	178943	366300	5
Q-4	58							
Q-1	59	610.5	104327	3.759	392211	364453	756664	16997
Q-2	59							
Q-3	59	848.5	149342	3.869	577733	544442	1122175	5253
0 - 4	59							
0 - 1	60	449.5	77857	4.465	347606	292362	639968	1150
Q-2	60						037700	1130
0-3	60	261-0	43829	5,126	224667	162053	386720	5
0-4	60	20100	13027	20 12 U	224001	102055	300120	· · ·
0-1	61	711.0	121423	4 611	550010	411157	071067	-12
0-2	61	11100		1. OIT	<i>JJJJ</i>	411174	911001	-16
0-3	61	466 5	84572	4 05 2	410012	204710	012522	443
0-4	61	1001	04712	4.772	410013	374110	813923	441
n = 1	62	454 5	112066	4 772	53/ 773	502704	1037550	,
0-2	42	000.0	112044	4.113	224112	202100	1031330	-4
Q-2	62	057 0	1(1012	( 200	(777)5	72/0/0	1/1/575	
Q-3	02	957.0	101012	4.209	0///15	130800	1414575	975
Q-4	02		1/7001		754 51 6			
Q-1	6.3	981.0	167581	4.323	724510	802664	1527174	53198
Q-2	63							
Q-3	63	811.0	136417	4.591	626304	714074	1340378	42977
Q-4	63							
Q-1	64	708.0	120710	4.750	573403	647706	1221109	29476
Q-2	64							
Q-3	64	450.0	79249	5.070	401800	457130	858930	7130
Q-4	64						,	
Q-1	65	3564.0	618027	3.927	2426847	3195298	5622145	9121
Q-2	65							
Q-3	65	273.0	45780	4.674	213995	302267	516262	45797
0-4	65							

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# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### 4-SYSTEM 5 5-SUBSYSTEM 50 GSE BASIC

	i IoM	NAN- I NTHS I	LABOR I HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR Matl
Q - 1 6	6 10	02.0	17826	3.354	59793	81690	141483	1150
Q-3 66	6 !	52.5	8810 3	3.834	33777	57270	91047	2931
TOTA	L 122	08.0 21	00305		9009000	9874823	18883823	216590

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### 4-SYSTEM 5 5-SUBSYSTEM 50 GSE BASIC

		MFG MATL	TOTAL MATERIAL	MPC	OTHER COST	SUB Total	G&A	TOTAL Cost
Q-1	58					56745		56745
Q-2	58							
Q-3	58		- 5			366305		366305
Q-4	58							
Q-1	59		16997	1379		775040		775040
Q-2	59							
Q-3	59		5253	445	4629	1132502		1132502
Q-4	59					·		
Q-1	60		1150	150	9366	650634	12397	663031
Q-2	60							
Q-3	60		5		3025	389750	7426	397176
Q-4	60							
Q-1	61		-12	-1	7379	978433	18182	996615
Q-2	61							
Q-3	61		441	37	4277	818278	15206	833484
Q-4	61							
Q-1	62	10523	10519	829	3063	1051969	17657	1069626
Q-2	62							
Q-3	62	166760	167735	13216	7687	1603213	26910	1630123
Q-4	62							
Q-1	63	328048	381246	37 55 2	6417	1952389	32627	1985016
Q-2	63							
Q-3	63	446157	489134	48179	6715	1884406	31507	1915913
Q-4	63				· · · · ·			
Q-1	64	275255	304731	32484	4632	1562956	33256	1596212
Q-2	64							
Q-3	64	110990	118120	42972	3114	1023136	21771	1044907
Q-4	64		1,X					
Q-1	65	74758	83879	25088	21	5731133	152906	5884039
Q-2	65							
Q-3	65	49653	95450	17028	157	628 <b>897</b>	16779	645676
0-4	65							

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 5 5-SUBSYSTEM 50 GSE BASIC

	MFG MATL	TOTAL MATERIAL	MPC	OTHER COST	SUB TOTAL	G & A	TOTAL Cost
Q = 1 66	45444	46594	9664	43	197784	5956	203740
Q-3 66	70861	73792	4538	79	169456	5103	174559
TOTAL	1578449	1795039	233560	60604	20973026	397683	21370709

## COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 5 5-SUBSYSTEM 51 GSE MAINTENANCE

		DESIGN	
		<b>/</b> ENGR	TOTAL
		HOURS	HOURS
		DOLLARS	DOLLARS
DESIGN/ENGINEERING		83812	83812
LABOR AT \$ 5.258		440678	440678
ENGR BURDEN AT \$	6.150	515432	515432
SHOP SUPPORT		21443	21443
LABOR AT \$ 4.450		95429	95429
TEST/QC		4891	4891
LABOR AT \$ 3.682	•	18007	18007
MFG BURDEN AT \$	4.837	127366	127366
ENGR MATERIAL		7532	7532
MPC		1865	1866
SUB-TOTAL		1206310	1206310
GEN & ADMIN		26637	26637
TOTAL COST		1232947	1232947

TIME-PHASED COST DETAIL-SEE PAGE II-538

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING

4-SYSTEM	5	
5-SUB SYSTEM	51 GSE	MAINTENANCE
SUBD OF WORK	DESIGN/ENGINEERING	

		MAN- MON THS	LABOR HOUR S	LABOR	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58	6.0	1074	4.397	4722	4882	9604
Q-2	58						
Q-3	58	27.0	4542	4.325	19643	17790	37433
Q-4	58						
Q-1	59	36.0	6189	4.276	26462	21375	47837
Q-2	59						
Q-3	59	24.0	4322	4.313	18640	15799	34439
Q-4	59						
Q-1	60		38	6.395	243	142	385
Q-2	60						
Q-3	60						
Q-4	60						
Q-1	61						
Q-2	61						
Q-3	61						
Q-4	61				~		
Q-1	62						
Q-2	62						
Q-3	62						
Q-4	62						
Q-1	63						
Q-2	63		-				
Q-3	63	39.0	6669	4.837	32257	29208	61465
Q-4	63						
Q-1	64	56.0	9598	3.427	32890	69206	102096
Q-2	64					•	
Q-3	64	66.0	11735	4.142	48612	73874	122486
Q-4	64						
Q-1	65	87.0	15092	5.662	85458	100019	185477
Q - 2	65						

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

DESIGN/ENGINEERING 4-SYSTEM 5 5-SUBSYSTEM 51 SUBD OF WORK DESIGN/ENGINEERING

	MAN- Months	LABOR HOURS	LABOR RATE	LABOR Dollars	BURDEN DOLL ARS	LABOR + Burden \$
Q-3 65	190.5	32030	5.006	160349	209981	370330
Q-1 66 Q-2 66	-100.5	-17414	3.231	-56264	-130859	-187123
Q-3 66	58.5	9937	6.809	57666	104015	171681
TOTAL	489.5	83812		440678	515432	956110

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SVS TEM	SHOP SUPPORT	ISE	MATNUENA NOT
5-SUB SYSTEM	5		THENTERMOL
SUBD OF WORK	DESIGN/ENGINEERI	NG	

		MAN- MONTHS	LABOR	LABOR	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q-1	58						
Q-2	58						
Q-3	53						
Q-4	58						
Q-1	59						
Q-2	59						
Q-3	59						
Q-4	59						
Q-1	60						
Q-2	60						
Q-3	60						
Q-4	60						
0-1	61						
Q-2	61						
Q-3	61			ĸ			
Q-4	61				1		
Q-1	62						
Q-2	62						
Q-3	62						
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	63	4.5	711	3.069	2182	2 960	5142
Q-4	63						
Q-1	64	4.5	859	3.217	2763	12599	15362
Q-2	64						· · <del>-</del>
Q-3	64	21.0	3734	3.554	13272	31776	45048
Q-4	64					·	
Q-1	65	19.5	3315	4.111	13627	16274	29901
0 - 2	65						

TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 5 GSE MAINTENANCE 5-SUBSYSTEM 51 SUBD OF WORK DESIGN/ENGINEERING

	MAN- MON THS	LABOR HOUR S	LABOR	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 65	33.0	5594	4.769	26676	26149	52825
Q-4 65 Q-1 66	27.0	4725	5,156	24363	25566	49929
Q-2 66 Q-3 66	15.0	2505	5.008	12546	12042	24588
TOTAL	124.5	21443		95429	127366	222795

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	TEST/QC	
4-SYSTEM	5 GSE	MAINTENANCE
5-SUBSYSTEM	51	
SUBD OF WORK	DESIGN/ENGINEERING	

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	63		10	3.100	31		31
Q-4	63						
0-1	64	10.5	1788	3.629	6489		6489
Q-2	64						
Q-3	64	16.5	2910	3.713	10804		10804
Q-4	64						
Q-1	65	1.5	139	3.705	515		515
Q-2	65						
0-3	65		33	3.909	129		129
Q-4	65						· · ·
Q-1	66		6	3.667	22		22
Q-2	66						
Q-3	66		5	3.400	17	•	17
тот	AL	28.5	4891		18007		18007

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

		MAN- MONTHS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
Q-1	58	6.0	1074	4.397	4722	4882	9604	
Q-2	58					*		
Q-3	58	27.0	4542	4.325	19643	17790	37433	
Q-4	58					•		
Q-1	59	36.0	6189	4.276	26462	21375	47837	
Q-2	59							
Q-3	59	24.0	4322	4.313	18640	15799	34439	
Q-4	59							
Q-1	60		38	6.395	243	142	385	
Q-2	60							
Q-3	60							
Q-4	60							
Q-1	61							
Q-2	61							
Q-3	61							
Q-4	ó1							
Q-1	62							
Q-2	62							
Q-3	62							
Q-4	62							
Q-1	63							
Q-2	63							
Q-3	63	43.5	7390	4.664	34470	32168	66638	262
Q-4	63							
Q-1	64	71.0	12245	3.442	42142	81805	123947	458
Q-2	64							
Q-3	64	103.5	18379	3.955	72688	105650	178338	751
Q-4	64							
Q-1	65	108.0	18546	5.370	99600	116293	215893	2381
Q-2	65			•				
Q-3	65	223.5	37657	4.970	187154	236130	423284	1357
Q - 4	65							

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	MAN- MONTHS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$	ENGR Matl
Q = 1 66 Q = 2 66	-73.5	-12683	2.514	-31879	-105293	-137172	1343
Q-3 66	73.5	12447	6.446	80229	116057	196286	<b>9</b> 80
TOTAL	642.5	110146		554114	642 <b>7</b> 98	1196912	7532

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

		MPC	SUB TOTAL	G&A	TOTAL COST
0-1	58		9604		9604
Q-2	58				
Q-3	58		37433		37433
Q-4	58				
Q-1	59		47837		47837
Q-2	59				
Q-3	59		34439		34439
Q-4	59		,		
Q-1	60		385	7	392
Q-2	60				
Q-3	60				
Q-4	60				
Q-1	61				
Q-2	61				
Q-3	61				
Q-4	61				
Q-1	62				
0-2	62				
Q-3	62				
Q-4	62				
Q-1	63				
Q-2	63				
Q-3	63	26	66926	1119	68045
Q-4	63		•		
Q-1	64	45	124450	2648	127098
0-2					
~ ~	64			2010	100001
Q-3	64 64	80	179169	3812	182 98 1
Q-3 Q-4	64 64 .64	80	179169	3812	182 98 1
Q-3 Q-4 Q-1	64 64 64 65	80 866	179169 219140	3812 5847	182 98 1 22498 7
Q-3 Q-4 Q-1 Q-2	64 64 64 65 65	80 866	179169 219140	5812	182 98 1 224987
Q-3 Q-4 Q-1 Q-2 Q-3	64 64 65 65 65	80 866 406	179169 219140 425047	3812 5847 11340	182 98 1 224 98 7 436 38 7

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	MPC	SUB TOTAL	G&A	TOTAL COST
Q = 1 66 Q = 2 66	240	-135589	-4083	-139672
Q-3 66	203	197469	5947	203416
TOTAL	1866	1206310	26637	1232947

# COST BREAKDOWNS B-70 AIRCRAFT STUDY

# 4-SYSTEM 5 GROUND SUPPORT EQUIPMENT

			DESIGN /ENGR HOURS DOLLARS	PROD Hours Dollars	TEST /QC HOURS DOLLARS	TOTAL Hours Dollars
DESIGN/ENGINEE	RING		1660546	3854	112	1664512
LABOR AT \$	4.407		7317658	16586	458	7334702
ENGR BURDEN	AT \$	4.852	8055532	20734	425	8076691
PRODUCTION			• All the second	224559		224559
LABOR AT \$	3.326			7467 98		746798
SHOP SUPPORT			293403		362	293765
LABOR AT \$	4.706		1381456		1040	1382496
PLANN ING				7858		7858
LABOR AT \$	3.633			28551		28551
TEST/QC			5605	14138	14	19757
LABOR AT \$	3.572		20717	49785	65	70567
MFG BURDEN	AT \$	4.471	1471631	967922	1377	2440930
ENGR MATERIAL			222090		2032	224122
MFG MATERIAL				1578449		1578449
MPC			30368	204948	110	235426
OTHER COST			60550	54		60604
SUB-TOTAL			18560002	3613827	5507	22179336
GEN & ADMIN			353723	70575	22	424320
TOTAL COST			18913725	3684402	5529	22603656

TIME-PHASED COST			
DETAIL - SEE PAGE II-548	IT-558	TT-564	TT-569

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING 4-SYSTEM 5 GROUND SUPPORT EQUIPMENT SUBD OF WORK DESIGN/ENGINEERING

		MÁN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q - 1 Q - 2	58 58	45.0	7436	4.371	32501	33826	66327
Q-3 0-4	58 58	295.5	49611	4.169	206320	196499	403319
Q-1 Q-2	59 59	516.0	88171	4.057	357 <b>7</b> 32	303156	660888
Q-3 Q-4	59 59	790.5	139094	4.000	556335	498123	1054458
0 - 1 0 - 2	60 60	439.5	76199	4.501	342970	285893	628863
Q-3 Q-4	60 60	261.0	43754	5.127	224326	162162	386488
Q-1 Q-2	61 61	711.0	121378	4.612	559767	410982	970749
Q-3 Q-4	61 61	465.0	84321	4.946	417049	394373	811422
Q-1 Q-2	62 62	568.5	97112	4.889	474778	445240	920018
Q-3 Q-4	62 62	82.5	13826	3.700	51159	82 978	134137
Q-1 Q-2	63 63	126.0	21483	3.967	85223	109934	195157
Q-3 Q-4	63 63	389.5	65409	4.943	323332	380113	703445
0-1 0-2	64 64	534.0	91206	5.113	466366	557661	1024027
Q-3 Q-4	64 64	370.5	65237	5.451	355594	428850	784444
Q-1 Q-2	65 65	3585.0	621290	3.970	2466474	3245264	5711738
ų-3	65	405.0	68072	5.005	340686	467419	808105

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING 4-SYSTEM 5 GROUND SUPPORT EQUIPMENT SUBD OF WORK DESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-4 65						
Q-1 66	-39.0	-6836	3.903	-26681	-86594	-113275
Q-2 66						
Q-3 66	82.5	13783	6.038	83227	139653	222880
TOTAL	9628.0	1660546		7317658	8055532	15373190

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### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# SHOP SUPPORT 4-SYSTEM 5 GROUND SUPPORT EQUIPMENT SUBD OF WORK DESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DCLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58 50		4	2.000	8	14	22
Q-2 Q-3	58 58		36	3.278	118	146	264
Q-4 Q-1	58 59	130.5	22157	2.725	60370	81977	142347
Q-2 Q-3	59 59	82.5	14425	2.748	39643	61617	101260
Q-4 Q-1 Q-2	59 60 60	9.0	1650	2.855	4710	6545	11255
Q-3	60 60					-386	-386
Q-4 Q-1	61 61						
Q = 2 Q = 3	61 61		11	2.455	27	42	69
0-4 0-1	62 ( 2	42.0	7117	5.006	35625	32.881	68506
Q-2 Q-3	62 62	<b>546.</b> 0	91772	4.996	458493	447996	906489
Q-4 Q-1	63 63	. 501.0	85585	5.237	448246	460787	909033
Q-3	63 63	246.0	41282	5.131	211812	211524	423336
Q-4 Q-1	63 64	13.5	2303	3.150	7255	19834	27089
Q-2 Q-3	64	25.5	4423	3.527	15598	35151	50 <b>7</b> 49
Q-4 Q-1	04 65	40.5	7014	3.752	26318	35192	61510
0-2	65 65	42.0	7019	4.468	31359	33058	64417

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# SHOP SUPPORT GROUND SUPPORT EQUIPMENT 4-SYSTEM 5 SUBD OF WORK DESIGN/ENGINEERING

	•	MAN- MON TH S	LABOR	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-4 Q-1	65 66	28.5	5002	5.102	25518	27215	52733
Q-2 ( Q-3 (	66 66	21.0	3603	4.540	16356	18038	34394
тоти	AL	1728.0	293403		1381456	1471631	2853087
4-SYSTEM

#### TIME PHASED EXPEND. **B-70 AIRCRAFT STUDY**

TEST/QC GROUND SUPPORT EQUIPMENT 5 SUBD OF WORK DESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLL ARS	LABOR + BURDEN \$
Q-3	58		4	2.250	9		9
Q-4	58						
0-1	59				-		
Q-2	59						
Q-3	59		26	2.846	74		74
Q-4	59		~				
Q-1 0 2	60		9	3.889	35		35
	<b>6</b> 0						
0-5	60						
0-1	61						
0-2	61						
0-3	61						
0-4	61						
0-1	62						
Q-2	62						
Q-3	62						
0-4	62						
Q-1	63		47	4.681	220		220
Q-2	63						
Q-3	63		36	4.806	173		173
Q-4	63	10 5					
Q-1	64	10.5	1858	3.615	6716		6716
Q=2	64 4.4	14 5	2041	3 707	10000		10000
0-4	0° <del>1</del> 64	10.0	2941	5.191	10403		10903
0-1	55	3.0	450	3.793	1707		1707
$\tilde{0}-2$	65	3.0	490	24122	<b>L</b> 101		LIVI
0-3	65	1.0	125	3.560	445		445
Q-4	65			100 m 20 100 100			
Q-1	66		35	4.229	148		148

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QCGROUND SUPPORT EQUIPMENT4-SYSTEM5SUBD OF WORK DESIGN/ENGINEERING

r	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-2 66 Q-3 66		74	3.878	287		287
TOTAL	31.0	5605		20717		20717

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 5 GROUND SUPPORT EQUIPMENT SUBD CF WORK DESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOURS	LABOR	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$	ENGR MATL
Q-1	58 50	45.0	7440	4.369	32509	33840	66349	
Q-2	20 50	205 5	10153		00/0/7			
0-4	59	292.3	49001	4.108	206947	196845	403592	
0-1	59	646.5	110328	3 700	419102	295122	002225	14091
0-2	59	01005	110320		410102	202123	003235	14201
$\overline{Q}$ - 3	59	873.0	153545	3.882	596052	559740	1155792	5254
Q-4	59			5.000	570052	0110	** >> * > ~	2223
Q-1	60	448.5	77858	4.466	347715	292438	640153	1144
Q-2	60							
Q-3	60	261.0	43754	5.127	224326	161776	386102	-1 6
Q-4	60							V.
Q-1	61	711.0	121378	4.612	55 9 <b>7</b> 6 <b>7</b>	410982	970749	-12
Q-2	61							
Q-3	61	465.0	84332	4.946	417076	394415	811491	441
Q-4	61							
	62	610.5	104229	4.897	510403	478121	988524	-4
Q-2	62	(20.5	105500	4	FOOLFO	F		
0-4	62	028.0	TUDDAR	4.020	208652	530914	1040626	475
Q <b>-</b>	62	627 0	107115	4 092	522400	570701	1106610	52100
0-2	63	02100	10/11.5	4.202	400666	210121	1104410	221.90
Q-3	63	635.5	106727	5-016	535317	591637	1126954	42220
0-4	63	€	100121	20010	1 2 4 5 6 6	271021	1120754	73233
0-1	64	558.0	95367	5.037	480337	577495	1057832	29934
Q-2	64							
Q-3	64	412.5	72601	5.263	382095	464001	846096	7881
Q-4	64							
Q-1	65	3628.5	628754	3.967	2494499	3280456	5774955	11502
Q-2	65							
0-3	65	448.0	75216	4.952	372490	500477	872967	47154
Q-4	65				·			
Q <b>-1</b>	66	-10.5	-1799	• 564	-1015	-59379	-60394	2493

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# GROUND SUPPORT EQUIPMENT

4-215	164	l	5
SUBD	GF	WORK	DESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOUR S	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$	ENGR Matl
Q-2 66 Q-3 66	103.5	17460	5.720	99870	157691	257561	3911
TOTAL	11387.0	1959554		8719831	9527163	18246994	222090

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

GROUND SUPPORT EQUIPMENT

SUBD OF WORK DESIGN/ENGINEERING

5

4-SYSTEM

		MPC	OTHER COST	SUB Total	GEA	TOTAL
Q-1	58			66349		66349
Q-2	58					
Q = 3	58			403592		403592
Q-4	58	12/0				
Q−1 0-2	27 50	1269		819485		819485
972 0-3	50	6.4 E	4420	11//120		11// 100
Q-5 0-4	50	440	4029	1100120		1106120
0-1	60	150	0366	450912	12400	442012
0-2	60	100	9300	020012	12400	003213
0-3	60		3025	389126	7414	396540
Q-4	50			20/120	3 7 4 7	570540
0-1	61	-1	7379	978115	18176	996291
Q-2	61					
Q-3	61	37	4277	816246	15168	831414
Q-4	61					
0-1	62		2608	991128	16636	1007764
Q-2	62					
Q-3	ó2	76	7757	1049434	17615	1067049
0 - 4	62					
Q-1	63	5240	5638	1168486	19520	1188006
Q-2	63					
0-3	63	4259	7851	1182303	19768	1202071
Q-4	63	<b></b>		1005540		1110000
0-2	04	5187	4010	1095569	23311	1118880
0-2	04 67	3676	2100	950750	19204	070057
0-4	64	2014	2100	029129	10294	010000
0-1	65	3594	17	5790068	154479	5944547
0-2	65	7774	11	5190000	174417	JTTTTT
0-3	65	8576	157	928854	24782	953636
0-4	65		<b>▲</b> ✓ •	· · · · · · · · · · · · · · · · · · ·		in the same and the same
Q-1	66	479	43	-57379	-1728	-59107

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> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

GROUND SUPPORT EQUIPMENT

4-SYSTEM 5 SUBD OF WORK DESIGN/ENGINEERING

	MPC	OTHER Cost	SUB Total	G & A	TOTAL COST
Q-2 66 Q-3 66	383	79	261 93 4	7888	269822
TOTAL	30368	60550	18560002	353723	18913725

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#### **APRIL 1972**

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## DESIGN/ENGINEERING 5 GROUND SUPPORT EQUIPMENT

SUBD OF WORK PRODUCTION

4-SYSTEM

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	62		8	3.000	24	34	58
Q-2	62						
Q-3	62	1.5	138	3.145	434	584	1018
Q-4	62						
Q-1	63		-36	3.333	-120	-21	-141
Q-2	63						
Q-3	63	9.0	1526	4.290	6547	7108	13655
Q-4	63						
ର−1	64	6.0	1138	4.651	5293	6730	12023
Q-2	64						
Q-3	64	3.0	401	3.736	1498	2257	3755
Q-4	64						
0 - 1	65	1.5	275	4.382	1205	1830	3035
Q-2	65						- · <b>-</b> -
Q-3	65	1.5	168	4.899	823	1085	1908
Q-4	65						
Q-1	66		113	3.920	443	644	1087
Q-2	66						
Q-3	66		123	3.569	439	483	922
тот	T AL	22.5	3854		16586	20734	37320

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### PRODUCTION 4-SYSTEM 5 GROUND SUPPORT EQUIPMENT SUBD OF WORK PRODUCTION

		MAN- MONTHS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	61	1.5	240	7.238	1737	295	2032
Q-4 Q-1 Q-2	61 62 62	40.0	6783	3.052	20699	24301	45000
Q-3 Q-4	62 62	309.0	52028	3.027	157511	204229	361740
Q-1 Q-2	63 63	342.0	58394	3.134	183018	231964	414982
Q-3 Q-4	63 63	188.5	31620	3.315	104820	140106	244926
Q-1 Q-2	64 64	189.0	32299	3.573	115404	139808	255212
Q-3 Q-4	64 64	124.5	21951	3.717	81602	94090	175692
Q-1 Q-2	65 65	36.0	6363	4.182	26613	27484	54097
Q-3 Q-4	65 65	39.0	6604	3.421	22590	34661	57251
Q-1 Q-2	66 66	31.5	5502	4.199	23102	32 705	55807
Q-3	<b>6</b> 6	16.5	2775	3.496	9702	14054	23756
TOT	AL	1317.5	224559		746798	943697	1690495

### APRIL 1972

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PLANNING

4-SYSTEM 5 SUBD OF WORK PRODUCTION

GROUND SUPPORT EQUIPMENT

	MAN- MON THS	LABOR	LABOR RATE	LABOR DOLLARS	BURDEN Doll Ars	LABOR + BURDEN \$
Q-1 62	4.5	751	3.688	2770	330	3100
$Q-2 \ 62$ $Q-3 \ 62$ $Q-4 \ 62$	7.5	1272	3.106	3951	1073	5024
Q-1 63 Q-2 63						
Q-3 63 Q-4 63	15.0	2616	4.043	10576	7391	17967
Q = 1 64 Q = 2 64	7.5	1156	3.548	4102	5478	9580
Q-3 64 Q-4 64	3.0	528	3.371	1780	2432	4212
Q-1 65 Q-2 65	1.5	386	3.443	1329	1821	3150
Q-3 65 Q-4 65	3.0	470	3.311	1556	2174	3730
Q-1 66 Q-2 66	3.0	449	3.666	1646	2427	4073
0-3 66	1.5	230	3.657	841	1099	1940
TOTAL	46.5	7858		28551	24225	52776

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## TEST/QC

4-SYSTEM 5 SUBD OF WORK PRODUCTION

GROUND SUPPORT EQUIPMENT

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	62	1.5	273	3.209	876		876
Q-2	62						
Q-3	62	12.0	1976	3.121	6167		6167
Q-4	62						
Q-1	63	12.0	2108	3.759	7923		7923
Q-2	63						
Q-3	63	7.5	1318	2.666	3514		3514
Q-4	63						
0-1	64	18.0	2995	3.475	10409		10409
Q-2	64						
0-3	64	12-0	2147	3.499	7513		7513
0-4	64				· · · · · · · · · · · · · · · · · · ·		
0-1	65	4.5	795	3-523	2801		2801
0-2	65		• • •				2001
0-3	65	6-0	979	3.769	3690		3690
0-4	65			50107	5050		5050
0-1	66	4-5	878	4.257	3738		3738
$\tilde{\Omega}$ - 2	66		010				
Q-3	66	4.5	669	4.714	3154		3154
тот	AL	82.5	14138		49785		49785

#### **APRIL 1972**

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. **B-70 AIRCRAFT STUDY**

	4-SYSTEM 5 SUBD OF WORK PRODUCTION			GROUND SUPPO	ORT EQUIPMEN	T			
		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	MFG Matl	
Q-3 0-4	61 61	1.5	240	7.238	1737	295	2032		
Q-1 Q-2	62 62	46.0	7815	3.118	24369	24665	49034	10523	
Q-3 Q-4	62 62	330.0	55414	3.033	168063	205886	373949	166760	
Q-1 Q-2	63 63	354.0	60466	3.156	190821	231943	422764	328048	
Q-3 Q-4	63 63	220.0	37080	3.383	125457	154605	280062	446157	
Q-1 Q-2	64 64	220.5	37588	3.597	135208	152016	287224	275255	
Q-3 Q-4	64 64	142.5	25027	3.692 '	92393	98 <b>7</b> 79	191172	110990	
0-1 0-2	65 65	43.5	7819	4.086	31948	31135	63083	74758	
Q-3 Q-4	65 65	49.5	8221	3.486	28659	37920	665 <b>79</b>	49653	
0-1 0-2	66 66	39.0	6942	4.167	28929	35776	64705	45444	
Q-3	66	22.5	3797	3.723	14136	15636	29772	70861	
<b>T</b> 01	AL	1469.0	250409		841720	988656	1830376	1578449	

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM	5	GROUND	SUPPORT	EQUIP MENT
SUBD OF WORK	PRODUCTION		÷1	

	МРС	OTHER COST	SUB TOTAL	G&A	TOTAL COST
Q-3 61			2032	38	2070
Q-4 61					
Q-1 62	829	455	60841	1021	61862
Q-2 62					
Q-3 62	13140	-70	553779	9295	563074
Q-4 62					
Q-1 63	32312	779	783 903	13107	797010
Q-2 63					
Q-3 63	43946	-1136	769029	12858	781887
Q-4 63					
Q-1 64	29342	16	591837	12593	604430
Q-2 64					
Q-3 64	40378	6	342546	7289	349835
0-4 64					
0-1 65	22360	4	160205	4274	164479
0-2 65					
Q-3 65	8858		125090	3337	128427
Q-4 65					
Q-1 66	9425		119574	3601	123175
0-2 66	•				
Q-3 66	4358		104991	3162	108153
TOTAL	204948	54	3613827	70575	3684402

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#### TIME PHASED EXPEND. **B-70 AIRCRAFT STUDY**

#### DESIGN/ENGINEERING

4-SYSTEM 5 SUBD OF WORK TEST/QC GROUND SUPPORT EQUIPMENT

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	60 ( 0		75	4.547	341	280	621
Q-4 Q-1	61		37	3.162	117	145	26.2
тот	AL		112		458	425	883

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### SHOP SUPPORT 4-SYSTEM 5 SUBD OF WORK TEST/QC GROUND SUPPORT EQUIPMENT

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	58 58		24	2.208	53	88	141
Q - 1 Q - 2	59 59	1.5	188	3.037	571	695	1266
Q-3	59 59		105	2.438	256	501	757
Q - 1	60 60		37	3.622	134	66	200
Q-3 0-4	60 60					-3	-3
Q-1	61		8	3.250	26	30	56
тот	AL	1.5	362		1040	1377	2417

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 5 SUBD OF WORK TEST/QC

GROUND SUPPORT EQUIPMENT

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q-3 59		14	4.643	65		65
TOTAL		14		65		65

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 5 SUBD OF WORK TEST/QC

GROUND SUPPORT EQUIPMENT

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
Q-3 58		24	2.208	53	88	141	5
Q-4 58 Q-1 59	1.5	188	3.037	571	695	1266	2016
Q=2 59 Q=3 59 Q=4 59		119	2.697	321	501	822	-1
Q = 1 60		37	3.622	134	66	200	6
Q-3 60		75	4.547	341	277	618	6
Q-1 61		45	3.178	143	175	318	
TOTAL	1.5	488		1563	1802	3365	2032

#### **APRIL 1972**

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	4-SYSTEI SUBD OF	M 5 WORK TES	T/QC	GROUND SUPPOR	r equipment
		MPC	SUB TOTAL	G & A	TOTAL COST
Q-3	58 50		146		146
Q-1	58 59	110	3392		3392
0-2 0-3	59 59		821		821
0-4 Q-1	59 60 .		206	4	210
Q-2 Q-3	60 60		624	12	636
Q-4 Q-1	60 61		318	6	324
างา	Γ AL	110	550 <b>7</b>	22	552 <b>9</b>

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING

4-SYSTEM

ON LHOINELNING

GROUND SUPPORT EQUIPMENT

		MAN-	LABOR	LABOR	LABOR	BUR DEN	LABOR +
		MONTHS	HOURS	RATE	DOLLARS	DOLLARS	BURDEN \$
Q-1	58	45.0	7436	4.371	32501	33826	66327
Q-2	58						
Q-3	58	295.5	49611	4.169	206820	196499	403319
Q-4	58						
Q-1	59	516.0	88171	4.057	357732	303156	660888
Q-2	59						
Q-3	59	790.5	139094	4.000	556335	498123	1054458
Q-4	59						
Q-1	60	439.5	76199	4.501	342970	285893	628863
Q-2	60						
Q-3	60	261.0	43829	5.126	224667	162442	387109
Q-4	60						
Q-1	61	711.0	121415	4.611	559884	411127	971011
Q-2	61					•	
Q-3	61	465.0	84321	4.946	417049	394373	811422
Q-4	61	,					
Q-1	62	568.5	97120	4.889	474802	445274	920076
Q-2	62						
Q-3	62	82.5	13964	3.695	51593	83562	135155
Q-4	62						
Q-1	63	126.0	21447	3.968	85103	109913	195016
0-2	63	200 5	icons	4			
Q-3	63	398.5	66935	4.928	329879	387221	/1/100
Q-4	63		00044	5 100	171/50	5(( )0)	102/050
	04	241+2	92344	2.108	411009	204391	1030020
	04	373 E	15/20	E (/ D	257002	421107	700100
0 4	.04	313.7	02030	<b>2 • 44</b> U	321092	451 107	100199
Q-4 0-1	04	3504 E	421545	2 070	2447470	2247004	571/772
	0) 45	. 5000+2	021000	2.910	2401013	5241074	5114115
₩-2 0-2	65	404 5	68240	5,005	341500	468504	810013
v−3 ∩_/	46	-100+J	00240	2002	241 JUS	<b>+UC JU</b>	010010
w,— •≠	00						

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	4- SY ST	[ FM 4								
			GROOMD POLICIT EGUT WEWL							
			ON-SITE LAB	IOR						
		MAN-	LABOR	LABOR	LABOR	BURDEN	LABOR +			
		MONTHS	HOURS	RATE	DOLLARS	DOLLARS	BURDEN \$			
2-1	66	-39.0	-6723	3.903	-26238	-85950	-112188			
2-2	66		·							
9-3	66	82.5	13906	6.017	83666	140136	223802			
тот	T AL	9650.5	1664512		7334702	8076691	15411393			

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PRODUCTION

4-SYSTEM 5

GROUND SUPPORT EQUIPMENT

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-3	61	1.5	240	7.238	1737	295	2032
Q-4	61						
Q-1	62	40.0	6783	3.052	20699	24301	45000
Q-2	62						
Q-3	62	309.0	52028	3.027	157511	204229	361740
Q-4	62						
Q-1	63	342.0	58394	3.134	183018	231964	414982
Q-2	63						
Q-3	63	188.5	31620	3.315	104820	140106	244926
Q-4	63						
Q-1	64	189.0	32299	3.573	115404	139808	255212
Q-2	64						
Q-3	64	124.5	21951	3.717	81602	94090	175692
Q-4	64						
Q-1	65	36.0	6363	4.182	26613	27484	54097
	05			<b>a</b> (a)	2250	~	67051
Q-3	07	39.0	6604	3.421	22590	34661	57251
0-1	60	<b>21 E</b>	5500	6 100	22102	22705	EE 007
0-2	00 44	01.0	2502	4.179	25102	52105	22001
Q-2	00 44	16.5	2775	2 404	0702	14054	22754
6-3	00	10+J	2115	2.470	9102	14024	23130
TOT	TAL.	1317.5	224559		746798	943697	1690495

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM

SHOP SUPPORT

## EM 5 GROUND SUPPORT EQUIPMENT

		MAN- MONTHS	LABOR	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58		4	2.000	8	14	22
Q-2	58						
Q-3	58		60	2.850	171	2.34	405
Q-4	58						
Q-1	59	130.5	22345	2.727	60941	82672	143613
Q-2	59						
Q-3	59	82.5	14530	2.746	39899	62118	102017
Q-4	59	_					
0-1	60	10.0	1687	2.871	4844	6611	11455
Q-2	60						
Q-3	60					-389	-389
Q-4	60						
Q-1	61		8	3.250	26	30	56
Q-2	61						
Q-3	61		11	2.455	27	42	69
Q-4	61				、 、		
Q-1	62	42.0	7117	5.006	35625	32881	68506
Q-2	62						
Q-3	62	546.0	91772	4.996	458493	447996	906489
Q-4	62						
ଦ-1	63	501.0	85585	5.237	448246	460787	909033
Q-2	63						
Q-3	63	246.0	41282	5.131	211812	211524	423336
Q-4	63					•	
Q-1	64	13.5	2303	3.150	7255	19834	27089
Q-2	64						
Q-3	64	25.5	4423	3.527	15598	35151	50749
Q-4	64						•
Q-1	65	40.5	7014	3.752	26318	35192	61510
Q-2	65						
Q-3	65	42.0	7019	4.468	31359	33058	64417
Q-4	65						

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM

C-11

SHOP SUPPORT GROUND SUPPORT EQUIPMENT

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q = 1	66 66	28.5	5002	5.102	25518	27215	52733
Q-3	66	21.0	3603	4.540	16356	18038	34394
тот	AL	1729.0	293765		1382496	1473008	2855504

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	PLANNING			
4-SYSTEM	5	GROUND	SUPPORT	EQUIPMENT

		MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1	62	4.5	751	3.688	2770	330	3100
Q-2	62						
Q-3	62	7.5	1272	3.106	3951	1073	5024
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	6.3	15.0	2616	4.043	10576	7391	17967
Q-4	63						
Q-1	64	7.5	1156	3.548	4102	5478	9580
Q-2	64						
Q-3	64	3.0	528	3.371	1780	2432	4212
Q-4	64						
Q-1	65	1.5	386	3.443	1329	1821	3150
Q-2	65						
Q-3	65	3.0	470	3.311	1556	2174	3730
Q-4	65						
Q-1	66	3.0	449	3.666	1646	2427	4073
Q-2	66						
Q-3	66	1.5	230	3.657	841	1099	1940
тот	AL	46.5	7858		28551	24225	52776

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 5

4-SYSTEM

GROUND SUPPORT EQUIPMENT

		MAN-	LABOR	LABOR	LABOR	<b>BUR DEN</b>	LABOR +
		MONTHS	HOURS	RATE	DOLLARS	DOLLARS	BURDEN \$
Q-3	58		4	2.250	9		9
Q-4	58						-
Q-1	59						
Q-2	59						
Q-3	59		40	3.475	139		139
Q-4	59						
Q-1	60		9	3.889	35		35
Q-2	60						
Q-3	60						
Q-4	60						
Q-1	61						
Q-2	61						
Q-3	61						
Q-4	61						
Q-1	62	1.5	273	3.209	876		876
Q-2	62						
Q-3	62	12.0	1976	3.121	6167		6167
Q-4	62						
Q-1	63	12.0	2155	3.779	8143		8143
Q-2	63						
Q-3	63	7.5	1354	2.723	3687		3687
Q-4	63						
Q-1	64	28.5	4853	3.529	17125		17125
Q-2	64						
Q-3	64	28.5	5088	3.619	18416		18416
Q-4	64	7 6			·		
Q = 1	.05	1.5	1245	3.621	4508		4508
Q-2	65		110/				
Q-3	0) (5	6.0	1104	3.145	4135		4135
Q-4	00	( )	012				
	00	6.0	913	4.256	3886		3886
V-2	<b>66</b>						

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM	TEST/QC 5	GROUND SUPPORT	EQUIPMENT
	ON-SITE LA	BOR	

LABOR + BURDEN \$	BURDEN DOLLARS	LABOR DOLLARS	LABOR RATE	LABOR HOUR S	MAN- MONTHS	
3441		3441	4.631	743	4.5	Q-3 66
70567		70567		19757	114.0	TOTAL

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### 4-SYSTEM 5 GROUND SUPPORT EQUIPMENT

		MAN- MONTHS	LABOR	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
Q-1	58	45.0	7440	4.369	32509	33840	66349	
Q-2 Q-3	58 58	295.5	49675	4.167	207000	196733	403733	5
Q-1 Q-2	50 59 59	646.5	110516	3.788	418673	385828	804501	16997
Q-3 Q-4	59 59	873.0	153664	3.881	596373	560241	1156614	5253
Q-1 Q-2	60 60	449.5	77895	4,466	347849	292504	640353	1150
Q-3 Q-4	60 60	261.0	43829	5.126	224667	162053	386720	5
Q-1 Q-2	61 61	711.0	121423	4.611	559 <b>91 C</b>	411157	971067	-12
Q-3 Q-4	61 61	466.5	84572	4.952	418813	394710	813523	441
Q-1 Q-2	62 62	656.5	112044	4.773	534772	502 <b>786</b>	1037558	-4
Q-3 Q-4	62 62	957.0	161012	4.209	677715	736860	1414575	975
Q-1 Q-2	63 63	981.0	167581	4.323	724510	802664	1527174	53198
Q-3 Q-4	63 63	855.5	143807	4.595	660774	746242	1407016	43239
Q - 1 Q - 2	64 64	780.0	132955	4.630	615545	729511	1345056	29934
Q-3 Q-4	64 64	555.0	97628	4.860	474488	562780	1037268	7881
Q-1 Q-2	65 65	3672.0	636573	3.969	2526447	3311591	5838038	11502
Q-3 0-4	65 65	496.5	83437	4.808	401149	538397	939546	47154
0-1	66	30.0	5143	5.428	27914	-23603	4311	2493

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 5 GROUND SUPPORT EQUIPMENT

	MAN- MON THS	LABOR	LABOR RATE	LABOR DOLLAR S	BURDEN Doll Ars	LABOR + BURDEN \$	ENGR MATL
Q-2 66 Q-3 66	126.0	21257	5.363	114006	173327	287333	3911
TOTAL	12857.5	2210451		9563114	10517621	20080735	224122

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### 4-SYSTEM 5 GROUND SUPPORT EQUIPMENT

		MFG	TOTAL		OTHER	SUB		TOTAL
		MATL	MATERIAL	MPC	COST	TOTAL	G&A	COST
Q = 1 Q = 2	58 58					66349		66349
Q-3	58 58		5			403738		403738
Q=4 Q=1	59		16997	1379		822877		822877
Q-2 Q-3	59 59		5253	445	4629	1166941		1166941
Q-4 Q-1	59 60		1150	150	9366	651019	12404	663423
Q-2 Q-3	60		5		3025	389 <b>7</b> 50	7426	397176
Q-4 Q-1	60 61		-12	- 1	7379	978433	18182	996615
Q-3	61		441	37	4277	818278	15206	833484
Q-4 Q-1	61 62	10523	10519	829	3063	1051969	17657	1069626
Q-3	62 62	166760	167735	13216	7687	1603213	26910	1630123
Q = 4 Q = 1	62 63	328048	381246	37552	6417	1952389	32627	1985016
Q-2 Q-3	63 63	446157	489396	48205	6715	1951 332	32626	1983958
Q-4 Q-1	64	275255	305189	32529	4632	1687406	35904	1723310
Q-2 Q-3	64 64	110990	118871	43052	3114	1202305	25583	1227888
Q-4 Q-1	64 65	74758	86260	25954	21	5950 <b>273</b>	158753	6109026
Q-3	65 65	49653	96807	17434	157	1053944	28119	1082063
Q-4 Q-1	05 66	45444	47937	9904	43	62195	1873	64068

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 5 GRCUND SUPPORT EQUIPMENT

	MFG	TOTAL MATERIAL	MPC	OTHER COST	SUB TOTAL	GEA	TOTAL COST
Q-2 66 Q-3 66	70861	74772	4741	79	366925	11050	377975
TOTAL	1578449	1802571	235426	60604	22179336	424320	22603656



#### SPARES WBS 6.0

#### Background

In conjunction with the approval of the master schedule revision W27R in late 1960, the authority to establish and begin procuring spare parts to support the XB-70 program was obtained. Limited activity had occurred prior to this time. This effort was restricted primarily to the study of spares requirements for the Weapon System 110A (RS-70) concept.

As a result of the program redirection the spares concept was drastically revised. Release of spares hardware requirements was not instigated until late 1960 - early 1961. During the intervening period between program redirection and early 1961, back-up or additional test units were being procured to support the engineering test activities. These units were not identified as spares.

In May 1964, at the time of Air Vehicle No. 1 roll-out, 13,743 air vehicle spare parts and 1,630 flight test ground support equipment spares had been identified.

Severe program funding limitations dictated that the XB-70 spares program be based on a "high risk" philosophy. This concept eliminated the procurement of "insurance type" items and emphasized the minimum procurement of high cost repairables while assuring maximum utilization of the available assets and production line support. Program emphasis was placed on expediting overhaul of repairables in contrast to procurement of additional units. In support of this concept, Air Vehicle No. 3 provided much of the spare parts utilized during the flight test portion of the program.

#### Cost Definition

Total recorded cost of \$18,010,222 includes all in-house costs associated with the definition, identification and fabrication of air vehicle and ground support equipment spare parts. The costs do not reflect any charges accumulated against Air Vehicle No. 3 which was utilized in the spares program after the vehicle was cancelled in early 1964.

Excluded from this WBS item are the costs identified by the subsystem contractors as "Spares." These costs are included in the subsystems. The following table provides a recap of the spares costs by subsystem.



1

		WBS 6.0
WBS	Title	Spares Cost*
1.1	Airframe Structures	<b>\$</b> 16,445
1.2	Environmental Control System	88,898
1.3	Propulsion System	344,210
1.4	Secondary Power System	2,025,156
1.5	Air Induction System	221,383
1.6	Flight Control System	70,430
1.8	Alighting and Arresting System	99,406
	Total	\$2,865,928

Refer to the individual WBS items for identification of the specific supplier providing the cost detail.

\* Cost reported do not include Material Procurement Cost or General and Administrative Burdens

6

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

CUDEVE

4-SYSTEM SPARES

		2-200212	
· .		60	TOTAL
		HOURS	HOURS
		DOLLARS	DOLLARS
DESIGN/ENGINEERING		478239	478239
LABOR AT \$ 3.516		1681648	1681648
ENGR BURDEN AT \$	5.172	2473632	2473632
PRODUCTION		735485	735485
LABOR AT \$ 3.132		2303896	2303896
PLANN ING		24444	24444
LABOR AT \$ 3.071	· .	75067	75067
TEST/QC		49983	49983
LABOR AT \$ 3.287		164304	164304
MFG BURDEN AT \$	4.137	3350652	3350652
ENGR MATERIAL		12960	12960
MFG MATERIAL		6357812	6357812
MPC		862366	862366
OTHER COST		348211	348211
SUB-TOTAL		17630548	17630548
GEN & ADMIN		379674	379674
TOTAL COST		18010222	18010222

SUBDI	VISION	OF WO	ORK	
COST	DETAIL	-SEE	PAGE	II <b>-</b> 584

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 6 5-SUBSYSTEM 60 SPARES BASIC

PROD HOURS DOLLARS	DESIGN /ENGR HOURS DOLLARS					
15994	462245		RING	IEE	NGI	ESIGN/E
57800	1623848		3.516	\$	ΑT	LABOR
62938	2410694	5.172	AT \$		DEN	NGR BUR
735485					ON	RODUCTI
2303896			3.132	\$	AT	LABOR
24444					•	ANNING
75067			3.071	\$	ΑT	LABOR
49983						EST/QC
164304			3.287	\$	AT	LABOR
3350652		4.137	AT \$		EN	G BURD
	12960			L	ERIA	NGR MAT
6357812				-	RIAL	G MATE
86 08 28	1538					20
8096	340115				ST	THER CO
13241393	4389155				L	JB-TOTA
289175	90499				MIN	EN & AD
13530568	4479654				ST	TAL CO
2 3 5 - 3	PROU HOURS DOLLARS 15994 5780( 62938 735485 2303898 24444 75067 49983 164304 3350652 6357812 860828 8096 13241393 289175 13530568	/ENGR PROU   HOURS HOURS   DOLLARS DOLLARS   462245 15994   1623848 57800   2410694 62938   735485 2303896   24444 75067   49983 164304   3350652 12960   12960 6357812   4389155 13241393   90499 289175   4479654 13530568	/ENGR PROU   HOURS HOURS DOLLARS   DOLLARS DOLLARS DOLLARS   462245 15994 1623848 57800   5.172 2410694 62938 735485 2303896   24444 75067 49983 164304   4.137 12960 6357812   1538 860826 340115 8096   4389155 13241393 90499 289175   4479654 13530566 13530566 13530566	$\begin{array}{rcrcrc} & /ENGR & PROT \\ HOURS & HOURS \\ DOLLARS & DOLLARS \\ \hline 00LLARS & DOLLARS \\ \hline$	$\begin{array}{rrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrrr$	/ENGR PROD   HOURS HOURS HOURS   DOLLARS DOLLARS DOLLARS   NG INEER ING 462245 15994   AT \$ 3.516 1623848 57800   DEN AT \$ 5.172 2410694 62938   ON 735485 2303896 24444   AT \$ 3.071 75067 49983   ST 12960 6357812   IS38 860826 340115 8096   L 4389155 13241393 93   MIN 90499 289175 13530566

TIME-PHASED COST			
DETAIL-SEE PAGE	II <b>-</b> 585	II-589	II <b>-</b> 595

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING

4-SYSTEM6SPARES BASIC5-SUBSYSTEM60SPARES BASICSUBDOFWORKDESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLL ARS	LABOR + BURDEN \$
Q-1	59	21.0	3517	3.442	12105	13538	25643
Q-2	59						
Q-3	59	168.0	29646	3.381	100239	118019	218258
Q-4	59						
Q-1	60	115.5	19919	3.941	78501	86395	164896
Q-2	60						
Q-3	60	34.5	5679	4.459	25322	22 893	48215
Q-4	60						
Q-1	61	204.0	34719	4.275	148427	138845	28 <b>7272</b>
Q-2	61						
Q-3	61	129.0	23461	3.659	85843	119167	205010
Q-4	61						
Q-1	62	136.5	23271	3.322	77299	98669	175968
Q-2	62						
Q-3	62	204.0	34219	3.103	106188	142157	248345
Q-4	62						
Q-1	63	261.0	44509	3.089	137487	204304	341791
Q-2	63						
Q-3	63	272.5	45741	3.155	144319	229120	373439
Q-4	63						
Q-1	64	267.0	45653	3.266	149104	240163	3892 <b>67</b>
Q-2	64						
Q-3	64	210.0	36888	3.479	128324	207198	335522
Q-4	64						
Q-1	65	205.5	35739	3.669	131117	229207	360324
Q-2	.65						
Q-3	65	175.5	29518	3.693	109015	191353	300368
Q-4	65						
Q-1	66	169.5	29341	3.780	110911	227 536	338447
n - 2	66						

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING

4-SYSTEM6SPARES BASIC5-SUBSYSTEM60SPARES BASICSUBDOFWORKDESIGN/ENGINEERING

LABOR + BURDEN \$	BUR DEN DOLL ARS	LABGR DOLLARS	LABOR RATE	LABOR HOURS	MAN- MONTHS	
221777	142130	79647	3.899	20425	121.5	Q-3 66
4034542	2410694	1623848		462245	2695.0	TOTAL

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SPARES BASIC

4-SYSTEM 6 82PA 5-SUBSYSTEM 60 SUBD OF WORK DESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOUR S	LABOR	LABOR DOLLARS	BURDEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
Q-1	59	21.0	3517	3.442	12105	13538	25 <b>643</b>	
Q-2	59							
Q = 3	59	168.0	29646	3.381	100239	118019	218258	
Q-4 0-1	59	116 6	10010	2 0/1	70501	01 205	1//00/	
0-2	60	110.0	19919	2.741	10201	00 292	104070	
0-3	60	. 34.5	5679	4.459	25222	22893	48215	
0-4	60	5105	2017	10122		22075	40212	
Q-1	61	204.0	34719	4.275	148427	138845	287272	
Q-2	61							
Q-3	61	129.0	23461	3.659	85843	119167	205010	372
Q-4	61					•		
Q-1	62	136.5	23271	3.322	77299	98 <b>66</b> 9	175968	253
Q-2	62							
Q-3	62	204.0	34219	3.103	106188	142157	2483 <b>45</b>	7021
Q-4	62	2(1.0	11500	2 220	107/07	201 201	2/1701	2201
V-1 0-2	63	201.0	44505	3.089	137487	204304	341791	2304
0-3	63	272.5	45741	3, 155	144319	229120	373439	162
0-4	63	21203	1714	2012	TTTOTO	227120	212422	IVE
0-1	64	267.0	45653	3.266	149104	240163	389267	557
0-2	64							
Q-3	64	210.0	36888	3.479	128324	207198	335522	79
Q-4	64							
Q-1	65	205.5	35739	3.669	131117	229207	360324	1676
<b>Q-</b> 2	65							
Q-3	65	175.5	29518	3.693	109015	191353	300368	259
Q-4	65							
Q-1	66	169.5	29341	3.780	110911	227536	338447	277
Q-2	66	101 5	20425	2 000	70747	143130	221777	
<u>y</u> -3	00	121.5	20420	2.848	19041	142130	221111	
TOT	TAL	2695.0	462245		1623848	2410694	4034542	12960
### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM6SPARES BASIC5-SUBSYSTEM60SUBDOFWORKDESIGN/ENGINEERING

		мрс	OTHER COST	SUB Total	G&A	TOTAL COST
0-1	59			25643		25643
Q-2	59			-		
0-4	59 50			218258		218258
n-1	60		1914	166710	2174	160006
Q-2	60		TOTA	100710		103000
$\bar{0} - \bar{3}$	60		377	48592	92.6	49518
Q-4	60					
Q-1	61		324 <b>7</b>	290519	5399	295918
Q-2	61					
Q-3	61	31	924	206337	3834	210171
Q-4	61					
Q-1	62	20	3234	179475	3012	182487
0-2	62	550	10777	275207	1120	270010
$Q^{-5}$	0Z 62	600	19577	215290	4020	279916
Q-1	63	227	41263	385585	6447	392032
Q-2	63					
Q-3	63	16	36073	409690	6850	416540
Q-4	63					
Q-1	64	59	29996	419879	8934	428813
Q-2	64					
Q-3	64	28	34572	370201	7877	378078
Q - 4	64	EOI	31353	201052	10500	×~~
0-2	65	501	34352	390853	10588	407441
Q = 2	65	46	37875	22956.9	0022	247590
6-4	65	10	21012	535,740	7032	541500
0-1	66	57	55258	394039	11867	405906
Q-2	66					
Q-3	66		41753	263530	7937	271467
тот	TAL.	1538	340115	4389155	90499	4479654

II**-**588

CVC TCM

TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

### DESIGN/ENGINEERING

4-212161	0	
5-SUB SYST	EM 60	SPARES BASIC
SUBD OF W	ORK PRODUCTIO	IN

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLL ARS	LABOR + BURDEN \$
Q-3	62	•	64	4.016	257	• 313	570
Q-4	62						
Q-1	63		102	3,588	366	589	955
Q-2	63						
Q-3	63	1.5	302	4.063	1227	1709	2936
Q-4	63						
Q-1	64	3.0	53 <b>7</b>	3.724	2000	2878	4878.
Q-2	64						
Q-3	64	3.0	600	3.880	2328	3823	6151
Q-4	64						
Q-1	65	3.0	432	3.694	1596	2030	3626
Q-2	65						
Q-3	65	18.0	2938	3.563	10469	11121	21590
Q-4	65						
Q-1	66	28.5	5027	3.523	17710	18180	35890
Q-2	66						
Q-3	66	36.0	5992	3.646	21847	22 295	44142
TOT	AL	93.0	15994		57800	62938	120738

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### PRODUCTION

4-SYSTEM 6 5-SUBSYSTEM 60 SUBD OF WORK PRODUCTION

#### SPARES BASIC

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
0-1 62	295.5	50531	2.960	149563	196 879	346442
Q-3 62	<b>1521.</b> 0	255447	3.046	778023	1032722	1810745
Q-4 62 Q-1 63	793.5	135421	3.074	416283	569572	985855
Q-2 63 Q-3 63	378.0	63592	3.524	224097	299878	5239 <b>75</b>
Q-4 63 Q-1 64	243.0	41397	3.289	136171	203517	339688
Q-2 64 Q-3 64	177.0	31145	3.484	108506	140943	249449
Q-4 64 Q-1 65	327.0	56566	3.510	198552	256773	455325
Q-2 65 Q-3 65	177.0	29613	3.363	99598	149807	249405
Q = 4 65 Q = 1 66	169.5	29336	1.645	48246	211038	259284
Q-2 65 Q-3 66	252.0	42437	3.413	144857	219101	363958
TOTAL	4333.5	735485		2303896	3280230	5584126

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SPARES BASIC

PLANNING 4-SYSTEM 6 5-SUBSYSTEM 60 SUBD OF WORK PRODUCTION

		MAN- MON THS	LABOR HOUR S	LABOR	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-3	61	1.5	294	2.707	796		796
Q-4	61						
Q-1	62	21.0	3693	2.706	9994		9994
Q-2	62						· · · ·
Q-3	62	31.5	5285	2.608	13785		13785
Q-4	62			•			
Q-1	63		11	3.727	41		41
Q-2	63						
Q-3	63	40.5	6837	3.651	24965	30063	55028
Q-4	63						
Q-1	64	9.0	1536	3.068	4713	7512	12225
Q-2	64						
Q-3	64	9.0	1505	2.950	4440	7 2 2 7	11667
Q-4	64						
Q-1	65	12.0	2140	3.043	6512	10100	16612
Q-2	65						
Q-3	65	7.5	1314	3.012	3 95 8	6045	10003
Q-4	65						2000
Q-1	66	6.0	1018	3.128	3184	5483	8667
Q-2	66						
Q-3	<b>6</b> 6	4.5	811	3.303	2679	3992	6671
тот	AL	142.5	24444		75067	70 422	145489

### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SPARES BASIC

#### TEST/QC

4-SYSTEM 6 5-SUBSYSTEM 60 SUBD OF WORK PRODUCTION

		MAN-	LABOR	LABOR	LABOR	BURDEN	LABOR +
		MONTHS	HOURS	RATE	DOLLARS	DOLL ARS	BURDEN \$
Q <b>-1</b>	62	9.0	1423	2.984	4246		4246
Q-2	62						
Q-3	62	69.0	11474	3.040	34883		34883
Q-4	62						
Q-1	63	51.0	8741	3.058	26733		26733
Q-2	<b>6</b> 3						
Q-3	63	33.0	5444	3.048	16593		16593
Q - 4	63						
Q-1	64	22.5	3779	3.326	12570		12570
0-2	64			_			
Q-3	64	15.0	2630	3.589	9439		9439
Q-4	64						
Q-1	55	30.0	5076	3.493	17732		17732
	65	070					
Q = 3	62	21.0	4483	3.408	15279		15279
Q=4 0-1	00 44	10.0	51/4	2 0/0	10070		10070
Q = 1 Q = 2	00 4 4	18.0	5144	3.840	12073		12073
0-2	00 44	<b>77 E</b>	7700	2 20 /	11751	·	· · · · · · · · · · · · · · · · · · ·
0-4	60	22.00	SI09	<b>&gt;•</b> 8₩4	14700		14756
W-4	1) (J						
тот	AL	297.0	49983		164304		164304

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

, SPARES BASIC

4-SYSTEM 6 5-SUBSYSTEM 60 SUBD OF WORK PRODUCTION

MAN-LABOR LABOR LABOR BUR DEN LABOR + MFG MON THS HOUR S RATE DOLLARS **BURDEN \$** DOLL ARS MATL Q-3 61 1.5 294 2.707 796 796 Q-4 61 Q-1 62 325.5 55647 2.944 163803 196879 360682 70552 Q-2 62 1621.5 Q-3 62 272270 3.037 826948 1033035 1859983 288871 Q-4 62 Q-1 63 844.5 . 144275 3.073 443423 570161 1013584 1597741 Q-2 63 Q-3 63 453.0 76175 3.504 266882 331650 598532 715023 0-4 63 277.5 422870 Q-1 64 47249 3.290 155454 213907 369361 0-2 64 Q-3 64 204.0 35880 3.476 124713 151993 276706 265679 Q-4 64 Q-1 65 3.494 372.0 64214 224392 268903 493295 261256 Q-2 65 Q-3 65 229.5 38348 3.372 129304 166973 296277 548220 0-4 65 Q-1 66 222.0 38525 2.108 1045141 81213 234701 315914 Q-2 66 Q-3 66 315.0 53029 3.472 184139 245388 429527 1142459 Q-4 66 TOTAL 4866.0 825906 2601067 3413590 6014657 6357812

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SPARES BASIC

4-SYSTEM 6 5-SUBSYSTEM 60 SUBD OF WORK PRODUCTION

		MPC	OTHER COST	SUB Total	GEA	TOTAL COST
Q-3	61		445	1241	23	1264
Q-4 Q-1	61 62	5559	3192	439985	7385	447370
Q-2 Q-3	62 62	22763	6381	2177998	36558	2214556
0-4 0-1 0-2	63 42	157377	-2035	2766667	46258	2812925
Q-3	63 42	70430	-5441	1378544	23049	1401593
Q = 1 Q = 2	64 64	45078	339	837648	17823	855471
Q-3 0-4	64 64	96654	1813	64085.2	13636	654488
0-1 0-2	65 65	78142	-1634	831 05 9	22173	853232
Q-3 Q-4	65 65	<b>978</b> 02	282	942581	25148	967729
Q-1 Q-2	66 66	216762	-1468	1576349	47475	1623824
Q-3 Q-4	66 66	70261	6222	1648469	49647	1698116
тот	AL	860828	8096	13241393	289175	13530568

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

### DESIGN/ENGINEERING

4-SYSTEM 6 5-SUBSYSTEM 60 SPARES BASIC

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-1	59	21.0	3517	3.442	12105	13538	25643
Q-2	59						
Q-3	59 50	168.0	29646	3.381	100239	118019	218258
Q=4	27 .	116 5	10010				
	00	112.2	19919	3.941	78501	86395	164896
Q-2	50						
Q = 3	60	34.5	5679	4.459	25322	22 893	48215
Q-4	60						
Q-1	61	204.0	34719	4.275	148427	138845	287272
Q-2	61						
Q-3	61	129.0	23461	3.659	85843	119167	205010
Q-4	61						
Q-1	62	136.5	23271	3.322	77299	98669	175968
Q-2	62						
Q-3	62	204.0	34283	3,105	106445	142470	248915
0-4	62						
Q-1	63	261.0	44611	3.090	137853	204893	342746
0-2	63			34070			512110
0-3	63	274.5	46043	3, 161	145546	230.829	376375
0-4	63	21102	10015	5.101	140040	230027	210010
0_1	64	270 0	46100	2 271	151104	242041	20/1/5
0-2	64	21040	40190	J • 2 + 1	171104	240041	
0-2	64	21.2 0	27/00	2 405	120452	271021	3/1/73
0-4	41	213.0	21400	3.405	120025	211021	241012
Q-4 0 1	04	200 F	2/171	2 ( ( 0	122712	201 227	2/2050
Q-1	05	208.7	30111	5.009	132113	231231	303950
Q-2	07	100 5	2015/	- · · · ·			
4-3	65	193.5	32456	3.681	119484	202474	321958
Q-4	65			_			
Q-1	66	198.0	34368	3.742	128621	245716	374337
0 - 2	66						

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

DESIGN/ENGINEERING 4-SYSTEM 6 5-SUBSYSTEM 60 SPARES BASIC

LABOR + BURDEN \$	BUR DEN DOLLARS	LABOR DOLLARS	LABOR RATE	LABOR HOURS	MAN- MONTHS	
265919	164425	101494	3.842	26417	157.5	0-3 66
4155280	2473632	1681648		478239	2788.5	TOTAL

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

### PRODUCTION 4-SYSTEM 6 5-SUBSYSTEM 60

SPARES BASIC

		MAN-	LABOR	LABOR	LABOR	<b>BUR DEN</b>	LABOR +
		MONTHS	HOURS	RATE	DOLLARS	DOLLARS	BURDEN \$
Q-1	62	295.5	50531	2.960	149563	196879	346442
Q-2 Q-3	62 62	1521.0	255447	3.046	779022	1022722	1010765
Q-4	62	192100	200111	<b>J</b> •0+0	110025	1032122	1810745
Q-1	63	793.5	135421	3.074	416283	569572	985855
0-2	63						
Q-3 Q-4	63 63	378.0	63592	3.524	224097	299878	523975
Q-1	64	243.0	41397	3.289	136171	203517	339688
Q-2	64						
Q-3	64	177.0	31145	3.484	108506	140943	249449
Q-4	64						
Q-1	65	327.0	56566	3.510	198552	256773	455325
Q-2	65						
Q-3	65	177.0	29613	3.363	99598	149807	249405
Q-4	65						
Q-1	66	169.5	29336	1.645	48246	211038	259284
Q-2	66						
Q-3	66	252.0	42437	3.413	144857	219101	363958
тот	AL	4333.5	735485		2303896	3280230	5584126

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PLANNING 4-SYSTEM 6 5-SUBSYSTEM 60 SPARES BASIC

		MAN- Months	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	61	1.5	294	2.707	796		796
Q-4	61						
Q-1	62	21.0	3693	2.706	9994		9994
Q-2	62						
Q-3	62	31.5	5285	2.608	13785		13785
Q-4	62						
Q-1	63		11	3.727	41		41
Q-2	63						
Q-3	63	40.5	6837	3.651	24965	30063	55028
Q-4	63						
Q-1	64	9.0	1536	3.068	4713	7512	12225
Q-2	64						
Q - 3	64	9.0	1505	2.950	444C	7227	11667
Q-4	64						
Q-1	65	12.0	2140	3.043	6512	10100	16612
Q-2	65						
Q-3	65	7.5	1314	3.012	3958	6045	10003
Q-4	65						
Q-1	66	6.0	1018	3.128	3184	5483	8667
Q-2	66						
Q-3	66	4.5	811	3.303	2679	3992	6671
TOT	AL	142.5	24444		75067	70422	145489

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 6 5-SUBSYSTEM 60 SPARES BASIC

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
0-1	62	9.0	1423	2.984	4246		4246
Q-3	62 62	69.0	11474	3.040	34883		34883
Q-4 Q-1	63 63	51.0	8741	3.058	26733		26733
Q-2 Q-3	63 63	33.0	5444	3.048	16593		16593
Q = 1 Q = 2	64 64	22.5	3779	3.326	12570		12570
Q-3 0-4	64 64	15.0	2630	3.589	9439		9439
Q-1 Q-2	65 65	30.0	5076	3.493	17732		17732
Q-3 Q-4	65 65	27.0	4483	3.408	15279		15279
Q-1 Q-2	66 66	18.0	3144	3.840	12073		12073
Q-3 Q-4	66 66	22.5	3789	3.894	14756		14756
тот	AL	297.0	49983		164304		164304

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM SPARES

		MAN- MON TH S	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR Matl
Q = 1 Q = 2	59 59	21.0	3517	3.442	12105	13538	25643	
Q-3 Q-4	59 59	168.0	29646	3.381	100239	118019	218258	
Q-1 Q-2	60 60	115.5	19919	3.941	78501	86 395	164896	
Q-3 Q-4	60 60	34.5	5679	4.459	25322	22 893	48215	
Q-1 Q-2	61 61	204.0	34719	4.275	143427	138845	287272	
Q-3 Q-4	61 61	130.5	23755	3.647	86639	119167	205806	372
Q-1 Q-2	62 62	462.0	78918	3.055	241102	295548	536650	253
0-3 0-4	62 62	1825.5	306489	3.045	933136	1175192	2108328	7021
Q-1 Q-2	6.3 63	1105.5	188784	3.077	580910	774465	1355375	2304
Q-3 Q-4	63 63	726.0	121916	3.373	411201	560770	971971	162
Q-1 Q-2	64 64	544.5	92902	3.278	304558	454070	758628	557
Q-3 Q-4	64 64	414.0	72768	3.477	253037	359191	612228	79
Q-1 Q-2	65 65	577.5	99953	3 <b>.</b> 55 <b>7</b>	355509	498110	853619	1676
Q-3 Q-4	65 65	405.0	67866	3.512	238319	358326	596645	259
Q-1 Q-2	66 66	391.5	67856	2.831	192124	462237	654361	277
Q-3 Q-4	66 66	436.5	73454	3.591	263786	387518	651304	
TOT	TAL	7561.5	1288151		4224915	5824284	10049199	12960

### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 6 SPARES

(

	MFG Matl	TOTAL MATERIAL	MPC	OTHER COST	SUB TOTAL	GξA	TOTAL Cost
Q = 1 59 Q = 2 59					25643		25643
Q-3 59 Q-4 59					218258		218258
Q-1 60 Q-2 60				1814	166710	3176	169886
Q = 3 60 Q = 4 60				377	48592	926	49518
Q = 1  61 Q = 2  61				3247	290519	5399	295918
2-3 61 0-4 61		372	31	1369	207578	3857	211435
Q-1.62 Q-2.62	70552	70805	5579	6426	619460	10397	629857
Q-3 62 Q-4 62	288871	295892	23316	25758	2453294	41178	2494472
Q-1 63 Q-2 63	1597741	1600045	157604	39228	3152252	52705	3204957
Q-3 63 Q-4 63	715023	715185	70446	30632	1788234	29899	1818133
Q-1 64 Q-2 64	422870	423427	45137	30335	1257527	26757	1284284
Q-3 64 Q-4 64	265679	265758	96682	36385	1011053	21513	1032566
Q-1 65 Q-2 65	261256	262932	78643	32718	1227912	32761	1260673
Q-3 65 Q-4 65	548220	548479	97848	38157	1281129	34180	1315309
Q-1 66 $Q-2$ 66	1045141	1045418	216819	53790	1970388	59342	2029730
Q-3 66 Q-4 66	1142459	1142459	70261	47975	1911999	57584	1969583
TOTAL	6357812	6370772	862366	348211	17630548	379674	18010222







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#### SPECIAL TEST EQUIPMENT WBS 7.0

#### Cost Definition

B-70 special test equipment is defined as that equipment which did not become an end product or part of an end product but was designed, fabricated and/or procured for use by the Manufacturing and Quality Control departments in the fabrication, testing, inspection and certification of the XB-70 systems or components during and after the manufacturing process. Examples of the types of equipment included in this WBS item are:

- a) continuity checkers
- b) analyzers
- c) electronic consoles
- d) hydraulic test units
- e) pressure test units
- f) leak testers
- g) other portable equipment

Cost data includes the identifiable expenditures to design, fabricate, test or procure the items listed above in support of the B-70 program. Some of the special test equipment was procured as capital investments and as such they are not included in the costs figures, Additional items were diverted from other in-house airframe programs and costs for this equipment is, likewise, absent from this WBS item. No costs for special test equipment developed or procured by the supplier for his internal use is included in the costs. See WBS 8.0, Tooling, for additional data.

Engineering special test equipment cannot be segregated from other in-house engineering costs. The costs for this engineering equipment are included in in the subsystem data.

### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 7 SPECIAL TEST EQUIPMENT

			5-SUBSYS	
			0	TOTAL
			HOURS	HOURS
			DOLLARS	DOLLARS
DESIGN/ENGINEER	ING		2348	2348
LABCR AT \$	4.201		9865	9865
ENGR BURDEN	AT \$	5.032	11814	11814
TOOLING AND STE			215320	215320
LABCR AT \$	3.598		774818	774818
TEST/QC			801	801
LABOR AT \$	3.532		2829	2829
MFG BURDEN	AT \$	4.207	909243	909243
TOOLING/STE MATE	_		596617	596617
мрс			61915	61915
SUB-TOTAL			2367101	2367101
GEN & ADMIN			44456	44456
TOTAL COST			2411557	2411557

SUBDIVISION OF WORK COST DETAIL - SEE PAGE II-605

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### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 7 5-SUBSYSTEM 0

SPECIAL TEST EQUIPMENT

- - - -	TOOLING AND STE HOURS DOLLARS	TOTAL Hours Dollars
DESIGN/ENGINEERING	2348	2348
LABOR AT \$ 4.201	9865	9865
ENGR BURDEN AT \$	5.032 11814	11814
TOOLING AND STE	215320	215320
LABOR AT \$ 3.598	774818	774818
TEST/QC	801	801
LABOR AT \$ 3.532	2829	2829
MFG BURDEN AT \$	4.207 909243	909243
TOOLING/STE MATL	596617	596617
MPC	61915	61915
SUB-TCTAL	2367101	2367101
GEN & ADMIN	44456	44456
TOTAL COST	2411557	2411557

TIME-PHASED COST DETAIL-SEE PAGE II-606

II**-**605

### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NASS-12100

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### TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

	DESIGN/	ENGIN	EERING		
4-SYSTEM	7				
5-SUB SYSTEM	0		SPECIAL	TEST	EQUIPMENT
SUBD CF WORK	TBOLING	AND	STE		

		MAN-	LABOR	LABOR	LABOR	<b>BURDEN</b>	LABOR +
		MUNTHS	HOURS	RATE	DOLLARS	DOLLARS	BURDEN \$
Q-1	61		46	3.565	164	156	320
Q-2	61						
Q-3	61		6	3.833	23	48	71
Q-4	61						
Q-1 (	62						
0-2	62						
Q-3 1	62		24	5.375	129	174	303
Q-4 (	62						
Q-1 (	63	1.5	378	5.595	2115	1995	4110
Q-2 (	63						
Q-3	63	6.0	955	3.074	2936	3831	6767
Q-4 1	63						
Q-1 d	64	3.0	420	5.455	2291	2591	4882
Q-2 (	64						
Q-3 (	64	1.5	170	4.894	832	883	1715
Q-4 (	64						
Q-1 (	65	1.5	180	4.406	793	1 1 9 1	1984
Q-2 (	65	1					
Q-3	65	1.5	135	3.356	453	693	1146
Q-4	65						
Q-1 (	66						
Q-2 (	66						
Q-3 (	66		34	3.794	129	252	381
тоти	AL	15.0	2348		9865	11814	21679

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### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TCOLING AND STE4-SYSTEM75-SUBSYSTEM0SUBD OF WORKTOOLING AND STE

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	60	1.5	149	2.953	440	567	1007
Q-4	60						
Q-1	61	7.5	1298	3.499	4542	4645	9187
Q-2	61						
Q-3	61	28.5	5229	4.039	21119	20092	41211
Q-4	61						
Q-1	62	28.5	4743	3.371	15988	17524	33512
Q-2	62						
Q-3	62	69.0	11615	3.080	35772	44575	80347
Q-4	62						
Q-1	63	39.0	6577	3.190	20981	24230	45211
Q-2	63						
Q-3	63	699.0	117475	3.514	412789	368082	780871
Q-4	63						
Q-1	64	55.5	9413	4.926	46364	43121	89485
Q-2	64						
Q-3	64	219.0	38514	3.345	128825	290653	419478
Q-4	64	,					
0-1	65	117.0	20204	3.757	75900	81813	157713
0-2	65						
Q-3	65		103	117.456	12098	13941	26039
тот	AL	1264.5	215320		774818	909243	1684061

### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	TEST/QC				
4-SYSTEM	7		SPECIAL	TEST	EQUTPMENT
5-SUB SYSTEM	0				- Cost Maint
SUBD OF WORK	TOOLING	AND	STE		

		MAN- MON THS	LABOR HOURS	RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
0-1	62	1.5	217	3.424	743		743
Q-2	62						
Q-3	62	1.5	320	3.425	1096		1096
Q-4	62						
Q-1	63		89	4.135	368		SAF
Q-2	63						
Q-3	63		71	3,141	223		223
0-4	63				2_ <b>C</b>		have been and
0-1	64		61	3-852	235		225
0-2	64				has not		San and San
0-3	64		23	4.000	92		02
0-4	64		Anna 144				7 *~
Q-1	65		15	3,533	52		5.2
0-2	65		т.,-		2		23
0-3	65		5	3.800	19		19
							<b>T</b> «
TOT	AL	3.0	801		2829		2829

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## 4-SYSTEM 7 SPECIAL TEST EQUIPMENT

		MAN-	LABOR	LABOR		BURDEN	LABOR +	TOOL/STE
		nontino	10005	NAIL	DULLARS	DULLARS	DUKUEN >	MAIL
Q-3	<b>6</b> 0	1.5	149	2.953	440	567	1007	1486
Q-4	<b>6</b> 0							1100
Q-1	61	7.5	1344	3.501	4706	4801	9507	21
Q-2	61							
Q-3	61	28.5	5235	4.039	21142	20140	41282	5420
Q-4	61							
Q-1	62	30.0	4960	3.373	16731	17524	34255	57699
Q-2	62							
Q-3	62	70.5	11959	3.094	36997	44749	81746	3717
Q-4	62		-				•	
<b>V-1</b>	63	40.5	7044	3.331	23464	26225	4968 <b>9</b>	18632
0-2	60	705 0	110501	2 512				
	62	102+0	118501	3.510	415948	371913	787861	436382
0-1	64	59 F	0004	4 04 1	40000	10710	04400	
0-2	64		2024	4.941	48890	42712	94602	50451
0-3	64	220-5	38707	3, 352	120740	201 524	4 21 29 5	040
Q-4	64	22009	50101	36 32 2	127147	271750	421205	742
Q-1	65	118.5	20399	3.762	76746	83.004	159750	16400
Q-2	65					00001	200100	10100.
Q-3	65	1.5	243	51.728	12570	14634	27204	5467
Q-4	65							
Q-1	66	•						
Q-2	66							
Q-3	66		34	3.794	129	252	381	
тот	AL	1282.5	218469		787512	921057	1708569	596617

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

### 4-SYSTEM 7 SPECIAL TEST EQUIPMENT

	MPC	SUB Total	GEA	TOTAL COST
Q-3 60	195	2638	51	2739
0-1 61 0-2 61	2	9530	177	9 <b>7</b> 07
Q = 3 61 Q = 4 61	458	47160	876	48036
9-1 62 9-2 52	4547	96501	1620	98121
Q-3 62 Q-4 62	293	85756	1439	87195
Q - 1  63 Q - 2  63	1835	70156	1173	71329
Q-3 63 Q-4 63	42984	1267227	21188	1288415
Q = 1 64 Q = 2 64	5378	150431	3201	153632
Q-3 64 Q-4 64	343	422570	899 <b>1</b>	431561
Q = 1 65 Q = 2 65	4905	181055	4831	185886
Q-3 65 Q-4 65	975	33646	898	34544
Q-1 66 Q-2 66				
2-3 66		381	11	392
TOTAL	61915	2367101	44456	2411557

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#### TOOLING WBS 8.0

This WBS item is segregated into three major categories of tooling. They are: dimensional tooling, contract supplemental tooling and handling equipment. Each category is discussed below.

#### Dimensional Tooling WBS Code 8.80

This WBS item contains all the in-house costs for the design, fabrication, assembly, inspection, installation, modification and maintenance of the jigs, dies, fixtures, molds, patterns, taps, gages and other equipment of a specialized nature defined as tooling articles to be used in the manufacturing of the B-70 parts and assemblies. Subcontractor identified costs for tooling are included in the subsystem cost data under the "Tooling and STE" Subdivisions of Work and do not appear in this item. In-house costs cannot be associated to a particular subsystem, therefore they are collected and displayed in this WBS block. Subcontractor identified tooling costs included in the subsystems are as shown below:

I	VBS	Title	Recorded Costs*
-	1.1	Airframe Structures	\$24,510,375
-	1.2	Environmental Control Subsystem	249,833
-	1.3	Propulsion Subsystem	179,269
Ţ	1.4	Secondary Power Subsystem	206,408
]	L.5	Air Induction Subsystem	31,404
ן	L.6	Flight Control Subsystem	467,552
נ	L.7	Personnel Accommodations and Escape Subsystem	m 58,354
ן	L.8	Alighting and Arresting Subsystem	1,851,888
		Total Cost	\$27,555,083

Refer to the individual subsystems for identification of the particular suppliers included in the totals.

As the result of redirection to three air vehicles and the severe funding limitations placed upon the program, a minimal tooling concept for limited rate usage was developed. Under this concept the primary cirteria for tool design were:

\* Excludes Material Procurement Costs and General and Administrative burdens

### II-611

#### SD72-SH-0003



WBS 8.0

- a) transportability
- b) demountability or capability of being repositioned without undue expense
- c) subject to partial dismantling for removal of parts assembled
- d) expandability to permit rework for possible rate changes.

No requirements for interchangeability or replaceability were imposed, however consideration was given to the replaceability problems for such items as elevons, tips, doors, etc.

A tooling concept team was established to identify the major assembly tooling breakdown and master model requirements. Additional responsibilities included the review and approval of all tool planning submitted by the subcontractors. Within the five major categories of tooling the following policies prevailed.

- a) Major Assembly Tooling -This included all floor type major assembly tools which were restricted to minimal fixturing within the tool. See Exhibit 2 page II-616 for a typical tool in this category.
- b) Master Tooling -

Master tools were provided in those areas required to coordinate the major assembly breakdown tools and master models for mold line control media. Masters fell in four categories: coordinate hole patterns and/or surfaces in more than one plane, coordinate hole patterns in contoured surfaces, coordinate hole patterns and/or surfaces for major cleavage points such as wing to fuselage shown on Exhibit 5 page II-619, and coordinate hole patterns in one plane due to engineering call-outs of matching holes requirements. North American supplied master tools to the subcontractors to control mating surfaces and attach patterns.

- c) Braze Tooling -Braze Fixtures (BRF), Prefit Assembly Jigs (PFAJ), Pre-Braze Apply Fixtures (PBAF) were provided for contoured panels as displayed by Exhibit 1, page II-615.
- d) Trim/Fusion Weld/X-Ray Equipment -Special fixed and portable tooling was provided to support the skate and skate track method of trimming, fusion welding and inspecting honeycomb panel joints. See Exhibit 4 page II-618, for a general arrangement of this type of equipment and Exhibit 3, page II-617 for further clarity of portable tools utilized on the skates and portable beams.



WBS 8.0

e) Detail Tooling -

Detail tooling was provided only in those areas where standard equipment or hand fabrication methods could not be employed or economically feasible. Sheet metal blanks or nibbler jigs (NJ) were ordered for blanks of .063 gauge and over and only in those cases where standard notching or nibbling was considered unsafe. Multiple type dies were provided to shape the various metals and to satisfy different forming requirements such as hot sizing and shrinkage considerations. Heat Treat Check Fixtures (HTCF) were provided for checking power hammered skins. Trim and drill tooling normally used after forming was not provided as these operations were performed on assembly. Machine parts were made by layout and setup wherever possible with tooling provided only in those cases where dimensional accuracy could not otherwise be achieved.

# Contract Supplemental Tooling WBS 8.81

Total costs of \$3,108,415 reflect all identifiable in-house costs to design, fabricate, assemble, procure, inspect, install, and maintain those items designated as contract supplemental tooling. These items are defined as those special or single purpose machines and devices such as erector set machines (excluding general purpose heads), jet engine starters and related gear boxes, sound abatement devices and other items as designated by the contract.

#### Handling Equipment WBS 8.82

Total costs of \$1,571,840 include all identifiable in-house costs incurred in the design and fabrication of special handling equipment. Supplier handling equipment is not included in this item. It is included in the vendor costs associated with the Tooling and STE subdivision of work assigned to the subsystem. Supplier handling equipment cannot be identified as a separate item. Handling equipment is grouped into the following categories for identification only. Costs are not available for these individual groups.

- a) Special Holding, Supporting and Material Handling Fixtures -This equipment was peculiar to and necessary for support of XB-70 parts during such operations as process cleaning and/or hot and cold sizing.
- b) Special Sheet and Skin Handling Equipment -These units facilitated the handling and transfer of light gauge honeycomb panel facing sheets in and out of stretchleveling operations.
- c) Special Dollies, Racks and Containers -This equipment was used primarily to store honeycomb panels, large coil and sheet metal, jigs, hot sizing tools, and to hold dies (up to 24 feet) during the cooling cycle from 1800°



WBS 8.0

6

downward prior to storage. In addition, special containers were utilized to store various parts between the sequential manufacturing operations. See Exhibit

d) Special Slings and Adaptors

Special slings were utilized in handling honeycomb core material on and off various machines, air vehicle components or assemblies (see Exhibit 7, page II-621 for a typical special sling), electrical blanket brazing fixtures, and apply fixtures. Special slings and/or adaptors were used on such items as the crew escape capsules, engines and their containers, fuselage panes, flaps, bulkheads, windshields and doors. See Exhibit 9, page II-623 which displays the lifting frame used in installing the accessory drive system.

- e) Special Skin Handling Trucks and Racks -Special trucks and racks with protective devices were used in the handling of large thin gauge stainless steel skins. (Exhibit 8, pg II-622)
- f) Special Floor Tracks Subject tracks were required in moving tooling in the heat treating/ plating operations plus on the assembly line to maintain alignment during mating operations.
- g) Special Work Platforms -Special platforms, fixed and moveable, were utilized in conjunction with the various assembly jigs, pick-up jigs, installation and checkout operations. See Exhibit 6, page II-620 displaying typical moveable maintenance platform.



MAJOR ASSEMBLY TOOLING (Crew Compartment Side Panel Assembly Jig)





SD72-SH-0003

EXHIBIT 3

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EXHIBIT<sup>4</sup> SD72-SH-0003

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EXHIBIT 5








#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

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#### 4-SYSTEM 8 TOCLING

	5-SUBSYS 0	5-SUBSYS 80	5-SUBSYS	5-SUBSYS 82
·	HOURS DOLLARS	HOURS DOLLARS	HOURS DOLLARS	HOURS DOLLARS
DESIGN/ENGINEERING	1887	70589	622	12502
LABOR AT \$ 3.436	6601	230082	2666	54755
ENGR BURDEN AT \$ 4.539	8358	317488	3279	59444
TOOLING AND STE	682470	6347169	234893	145072
LABOR AT \$ 3.412	2266425	21725638	798065	493442
PLANN ING	571106	1590		
LABOR AT \$ 3.504	2002107	4716		
TEST/QC	20487	274195	4593	2135
LABOR AT \$ 3.422	70716	938690	14964	7163
MFG BURDEN AT \$ 3.044	2568949	21250021	839884	553355
TOOLING/STE MATL	844371	22268688	1275734	342074
MPC	99362	1940267	117271	33624
OTHER COST	85	260953		
SUB-TOTAL	7866974	68936543	3051863	1543857
GEN & ADMIN	143367	1243123	56552	2 <b>7</b> 983
TOTAL COST	8010341	70179666	3108415	1571840

SUBDIVISION OF WORK		•		
COST DETAIL - SEE PAGE	II <b>-</b> 626	II-635	II <b>-</b> 647	II <b>-</b> 653

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

#### 4-SYSTEM 8 TOOLING

	TOTAL
	HUUKS
	DULLARS
DESIGN/ENGINEERING	85600
LABOR AT \$ 3.43	6 294104
ENGR BURDEN AT	<b>\$ 4.539 388569</b>
TOOLING AND STE	7409604
LABOR AT \$ 3.41	2 25283570
PLANN ING	572696
LABOR AT \$ 3.50	4 2006823
TEST/QC	301410
LABOR AT \$ 3.42	2 1031533
MFG BURDEN AT	\$ 3.044 25212209
TOOLING/STE MATL	24730867
MPC	2190524
OTHER COST	261038
SUB-TOTAL	81399237
GEN & ADMIN	1471025
TOTAL COST	82870262

#### SUBDIVISION OF WORK COST DETAIL - SEE PAGE II-659

### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 8 5-SUBSYSTEM 0

#### TOOLING

	TODLING AND STE HOURS DOLLARS	TOTAL Hours Dollars
DESIGN/ENGINEERING	1887	1887
LABOR AT \$ 3.498	6601	66 01
ENGR BURDEN AT \$ 4.	.429 8358	8358
TOOLING AND STE	682470	682470
LABOR AT \$ 3.321	2266425	2266425
PLANNING	571106	571106
LABOR AT \$ 3.506	2002107	2002107
TEST/QC	20487	20487
LABOR AT \$ 3.452	70716	70716
MFG BURDEN AT \$ 2.	.016 2568949	2568949
TOOLING/STE MATL	844371	844371
MPC	99362	99362
OTHER COST	85	85
SUB-TOTAL	7866974	7866974
GEN & ADMIN	143367	143367
TOTAL COST	8010341	8010341

TIME-PHASED COST DETAIL-SEE PAGE II-627

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## DESIGN/ENGINEERING

4-SYSTEM8TOOLING5-SUBSYSTEM0SUBD OF WORKTOOLING AND STE

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	60	1.5	340	5.676	1930	1278	320.8
Q-2	60						5200
Q-3	60	3.0	558	3.344	1866	2038	3004
Q-4	60				and the tar tar		
Q-1	61		15	2.267	34	147	191
Q-2	61				51	1.41	101
Q-3	61	1.5	198	2.071	410	837	1247
Q-4	51					. 00,	1271
Q-1	62	1.5	277	2.960	820	1 277	2097
Q-2	62			· - · · · -			2.071
Q-3	62	1.5	135	2.919	394	744	1128
Q-4	62						x130,
Q-1	63		110	2,955	325	595	920
Q-2	63						120
Q-3	63	1.5	162	2.846	461	875	1336
Q-4	63						1.000
Q-1	64		51	2.510	128	360	488
Q-2	64						
Q-3	64		21	5.762	121	59	180
Q-4	64						
Q-1	65		15	3.333	50	77	127
Q-2	65					••	
Q-3	65		5	12.400	62	71	133
TOT	AL	10.5	1887		6601	8 3 5 8	14959

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	TOOLING	AND	STE	
4-SYSTEM	8			TOOT THE
5-SUB SYSTEM	0			TOOLING
SUBD OF WORK	TOOLING	AND	STE	

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	60	21.0	3633	2.847	10344	11324	21668
Q-2	60						
Q-3	60	78.0	12996	2.857	37125	39952	77077
Q-4	60						
Q-1	61	163.5	27867	2.918	81326	-20088	61238
Q-2	61						
Q-3	61	769.5	139538	3.626	505949	488793	994742
Q-4	61						
Q <b>-1</b>	62	984.0	167958	3.248	545444	591396	1136840
0-2	62						
Q-3	62	399.0	66926	3.083	206359	262091	468450
Q-4	62						
Q-1	63	127.5	21.671	4.781	103619	110225	213844
Q-2	63						
Q-3	63	553.5	92954	2.923	271729	23356 <b>7</b>	505296
Q-4	63						
Q-1	64	378.0	64396	6.389	411399	538388	949787
Q-2	64						
Q-3	64	438.0	77207	.967	74673	91863	166536
Q-4	64						
Q-1	65	30.0	5324	2.255	12003	26635	38638
Q-2	65						
Q-3	65	12.0	2000	3.228	6455	9250	15705
тот	AL	3954.0	682470		2266425	2383396	4649821

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PLANNING 4-SYSTEM 8 TOOLING 5-SUBSYSTEM 0 SUBD OF WORK TOOLING AND STE

		MAN- MON TH S	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q-1	61	713.5	121729	3.174	386402	48 895	435297
Q-2	61						
Q-3	61	343.5	62211	3.069	190899	26357	217256
Q-4	61						
Q-1	62		7	•142	-1		-1
Q-2	62				,		
Q-3	62						
Q-4	62						
Q-1	63						
Q-2	63						
0-3	63	2229.0	374566	3.682	1379106	64642	1443748
Q-4	63	~					
Q-1	64	36.0	6021	2.550	15354	16464	31818
Q-2	64						
Q-3	64	10.5	1963	7.200	14134	13451	27585
Q-4	64	-					
Q-1	65	19.5	3351	3.297	11047	9138	20185
Q-2	65						
Q-3	65	6.0	1123	4.185	4700	6016	10716
Q-4	65						
Q-1	66		3				
Q-2	66						
Q-3	66	1.5	132	3.530	46.6	590	1056
тот	AL	3359.5	571106		2002107	185553	2187660

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TOOLING

TEST/QC 4-SYSTEM 8 5-SUBSYSTEM 0 SUBD OF WORK TOOLING AND STE

#### ON-SITE LABOR

		MAN- MONTHS	LABOR	LABOR RATE	LABOR DGLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	60		4	3.000	12		12
Q-2	60						
Q-3	60	1.5	219	3.306	724		724
()-4	60						
Q-1	61	3.0	458	3.264	1495		1495
Q-2	61						
Q-3	61	15.0	2813	3.194	8984		8984
0-4	51						
0-1	62	31.5	5376	3.319	17842		17842
Q-2	62						
Q <del>+</del> 3	62	15.0	2486	3.424	8512		8512
Q-4	62						
Q-1	63	7.5	1356	3.406	4619		4619
Q-2	63						
Q-3	63	18.0	3061	3.517	10766		10766
Q-4	63						
Q-1	64	18.0	2955	3.727	11013		11013
ପ-2	64						
Q-3	64	7.5	1319	3.854	5084		5084
Q-4	64						
0-1	65	3.0	440	3.784	1665		1665
TUT	AL	120.0	20487		70716		70716

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TOOLING

4-SYSTEM 8 5-SUBSYSTEM 0 SUBD OF WORK TOOLING AND STE

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR Dollars	BUR DEN DOLL ARS	LABOR + BURDEN \$	TOOL/STE MATL
0 - 1 0 - 2	60 60	22.5	3977	3.089	12286	12602	24888	1448
Q-3 Q-4	<b>6</b> 0	82.5	13773	2.884	39715	41 990	81705	5145
Q-1 Q-2	61 61	880.0	150069	3.127	469257	28954	498211	11324
Q-3 0-4	61 61	1129.5	204760	3.449	706242	515987	1222229	133786
Q-1 0-2	62 62	1017.0	173618	3.249	564105	592673	1156778	159401
Q-3	62 62	415.5	69547	3.095	215265	262835	478100	139404
0 - 1 0 - 2	63 63	135.0	23137	4.692	108563	110820	219383	64397
Q-3	63 63	2802-0	470743	3.531	1662062	299084	1961146	59319
Q-1 Q-2	64 64	432.0	73423	5.964	437894	555212	993106	172072
Q-3	64 64	456.0	80510	1.168	94012	105373	199385	75170
$\rho = 1$	65 65	52.5	9130	2.712	24765	35850	60615	15110
Q-3	65 65	18.0	3128	3.586	11217	15337	26554	7320
$Q^{-1}$	66 66		3					475
Q = 2 Q = 3	66	1.5	132	3.530	466	590	1056	
тот	AL	7444.0	1275950		4345849	2577307	6923156	844371

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### TOOLING

4-SYSTEM 8 5-SUBSYSTEM G SUBD OF WORK TOOLING AND STE

		мрс	OTHER COST	SUB TOTAL	GEA	TOTAL COST
Q-1	60	190	85	26611	507	27118
Q-2	60					
Q-3	60 60	677		87527	1668	89195
Q-4	69					
Q−1 0 2	01	166		510492	9486	519978
0-3	61	11205		1367320	25408	1207720
Q-4	61	******			62700	1392120
Q-1	62	12561		1323740	22302	1351042
Q-2	62					
Q-3	62	10985		628489	10549	639038
Q-4	62					
Q-1	63	5764		289544	4841	294385
Q-2	63	~				
Q-3	63	5309		2025774	33871	2059645
0.1	60	10262		1100501	0.51.0.0	1000 700
(-2)	64	10343		1183521	25182	1208703
0-3	64	27347		301902	6424	308326
Q-4	64	Approx 2 - 4 - 4 - 4			5121	
Q-1	65	4519		80244	2141	82385
Q-2	65					
Q-3	65	1306		35180	939	36119
Q-4	65					
Q-1	66	99		574	17	591
Q-2	66					
Q-3	66			1056	32	1088
тот	TAL	99362	85	7866974	143367	8010341

OF POOR QUALITY

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TOOLING

4-SYSTEM 8 5-SUBSYSTEM 0

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGI Mati
Q-1	60	22.5	3977	3.089	12286	12602	24888	
Q-2	60							
Q-3	60	82.5	13773	2.884	39715	41990	81705	
()-4	60							
Q-1	61	880.0	150069	3.127	469257	28954	498211	
Q-2	61		<b>.</b>					
Q-3 0-4	61	1129.5	204760	3.449	706242	515987	1222229	
0 - 1	62	1017.0	173618	3.249	564105	592673	1156778	
$\hat{\mathbf{Q}}$ - 2	62							
Q-3	62	415.5	69547	3.095	215265	262835	478100	
Q-4	62							
Q-1	63	135.0	231 <b>37</b>	4.692	108563	110820	219383	
Q-2	63							
Q-3	63	2802.0	470743	3.531	1662062	299084	1961146	
Q-4	63		<b></b>	<b>.</b>				
Q-1	64	432.0	73423	5.964	437894	555212	993106	
Q-2	64							
Q-3	64	456.0	80510	1.168	94012	105373	199385	,
0-4	64	r			•	0 - 0 - 0		
Q-1	65 (F	52.5	9130	2.712	24765	35850	60615	
0-3	C 2 65	18.0	3129	3 586	11217	15237	26554	
0-4	65	10.0	2160	J. 200	11211	10001	20004	
0-1	66		3					
$\hat{\mathbf{Q}} - \hat{\mathbf{Z}}$	66		-					
Q-3	66	1.5	132	3.530	466	590	1056	
тот	AL	7444.0	1275950		4345849	257730 <b>7</b>	6923156	

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TOOLING

4-SYSTEM 8 5-SUBSYSTEM 0

		TOOL/STE MATL	TOTAL MATERIAL	МРС	OTHER COST	SUB TOTAL	GEA	TOTAL Cost
0-1 0-2	60 60	1448	1448	190	85	26611	507	27118
Q-3 0-4	<b>6</b> 0	5145	5145	67 <b>7</b>		87527	1668	89195
Q-1 Q-2	61 61	11324	11324	<del>9</del> 5 <b>7</b>		510492	9486	519978
Q-3 Q-4	61 61	133786	133786	11305		1367320	25408	1392728
0-1 0-2	62 62	159401	159401	12561		1328740	22302	1351042
Q-3 Q-4	62 62	139404	139404	10985		628489	10549	639038
0-1 0-2	63 63	64397	. 64397	5764	,	289 <b>544</b>	4841	<b>294</b> 385
Q-3 Q-4	63 63	59319	59319	530 <b>9</b>		2025774	33871	2059645
Q - 1 Q - 2	64 64	172072	172072	18343		1183521	25182	1208703
Q-3 Q-4	64 64	75170	75170	2734 <b>7</b>		301902	6424	308326
Q-1 Q-2	65 65	15110	15110	4519		80244	2141	82385
Q-3 0-4	65 65	7320	7320	1306		35180	939	36119
Q-1 Q-2	66 66	475	475	99		574	17	591
Q-3	66					1056	32	1088
тот	AL	844371	844371	9936 <mark>2</mark>	85	7866974	143367	80 <b>1034</b> 1

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#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

		TOOLING	
		AND STE	TOTAL
		HOURS	HOURS
		DOLLARS	DOLLARS
DESIGN/ENGINEERING		70589	70589
LABOR AT \$ 3.259		230082	230082
ENGR BURDEN AT \$	4.498	317488	317488
TOOLING AND STE		6347169	5347169
LABOR AT \$ 3.423		21725638	21725638
PLANN ING		1590	1590
LABOR AT \$ 2.966		4716	4716
TEST/QC		274195	274195
LABOR AT \$ 3.423		938690	938690
MFG BURDEN AT \$	3.209	21250021	21250021
TOOLING/STE MATL		22268688	22268688
MPC		1940267	1940267
OTHER COST		260953	260953
SUB-TOTAL		68936543	68936543
GEN & ADMIN		1243123	1243123
TOTAL COST		70179666	70179666

•	TIME-PHASEI	COST	
	DETAIL-SEE	PAGE	IF-636

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

DESIGN/ENGINEERING

4-SYS	S TE!	4	8			DIMENCIA	
5-SUE	3 SY 3	STEM	80			DIMENSIONAL	TOOLING
SUBD	OF	WORK	TOOLING	AND	STE		

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR Doll Ars	BUR DEŃ DOLL ARS	LABOR + BURDEN \$
Q-1	58	1.5	152	4.704	715	690	1405
Q-2	58						
Q-3	58	1.5	308	3.078	948	1168	2116
Q - 4	58						
Q-1	59		3	5.333	16	13	29
0-2	59						
Q-3	59	10.5	1898	3.065	5818	6683	12501
Q-4	59						
Q-1	60	16.5	2877	2.506	7211	10837	18048
Q-2	60						
Q-3	60	24.0	4011	2.851	11437	15289	26726
Q-4	60						
Q-1	61	81.0	13916	2.794	38888	51822	90710
Q-2	61						
Q-3	61	93.0	16784	3.070	51535	70609	122144
0-4	61						
Q-1	62	46.5	8059	3.399	27396	37108	64504
0-2	62	~ / ^	0000	2 250			-
Q-3	62	54.0	8388	3.358	30179	46123	75302
Q-4	62 ()			( 110	10/07	01/71	4 4 1 1 2
Q-1	03	27.0	4121	4.112	19430	24010	44112
Q-2	63	20 5	1170	1 202	20517	25424	15050
Q-3	60	28.9	40 I V	4.373	20010	20430	42922
0-4	63	10 5	2200	<b>2 073</b>	12017	33474	25200
Q-2	04 67	19+0	POCC	2.013	12810	22414	55290
0-2	64		750	2 140	2255	2425	5700
0-6	64	4+2	150	<b>2014</b> 0	2000	3423	5100
0-1	04 65		100	5 651	616	851	1467
0-2	65		103	2001	010	071	1 10 1

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING

4-SYSTEM85-SUBSYSTEM80SUBD OF WORKTOOLING AND STE

	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 65		28	7.143	200	284	48.4
TOTAL	408.0	70589		230082	317488	547570

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TOOLING AND STE

4-SYSTEM	8			
5-SUB SYSTEM	80		DIMENSIONAL	TOOLING
SUBD OF WORK	TOOLING	AND	STE	

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-1	58	58.5	9756	3.818	37248	25807	63055
Q-2	58						
Q-3	58	31.5	5241	6.150	32232	9793	42025
Q-4	58						
Q-1	59	121.5	20749	4.487	93095	58510	151605
Q-2	59						
Q-3	59	451.5	79372	3.728	295909	237465	533374
Q-4	59				•		
Q-1	60	2073.0	359234	3.446	1237896	1071343	2309239
Q-2	60						
Q-3.	60	2413.5	405544	3.402	1379689	1175131	2554820
Q-4	60						
Q-1	61	8557.5	1460483	3,561	5201231	4331437	9532668
$\hat{Q} - 2$	61						
Q - 3	61	13525.5	2452558	3.299	8090511	8089157	16179668
0-4	61						
0-1	62	4243.5	724250	3.355	2429801	2640330	5070131
$\hat{Q}$ -2	62						
$\tilde{Q}-3$	62	1636.5	275009	3,373	927508	1195452	2122960
0-4	62						
$\bar{0} - 1$	63	1493.5	254854	3.287	837663	955152	1792815
$\bar{0}-2$	63						
0-3	63	-3.0	-425	132,215	56192	369804	425996
0-4	63						
0-1	64	1619.5	276354	3.402	940207	879221	1819428
$\overline{0-2}$	64			55.02			
$\tilde{u}$ -3	64	81.0	14371	7.681	110385	167202	277587
0-4	64						
0-1	65	42-0	7345	7.049	51777	42057	93834
Q-2	65	- <b>-</b>			~~	1 L	್ ಬ್ಯಾಂಗ್ ಗ್ರಾಂಗಿಯಲ್ 🧃

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TOOLING AND STE 4-SYSTEM 8 5-SUBSYSTEM 80 DIMENSIONAL TOOLING SUBD OF WORK TOOLING AND STE

		MAN- MONTHS	LABOR	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	65	15.0	2422	1.612	3905	1918	5823
Q-4	65						,
Q-1	66		4	750	3	20	23
Q-2	66						
Q-3	66		48	8.042	386		386
тот	AL .	36360.5	6347169	-	21725638	21249799	42975437

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PLANNING 4-SYSTEM 8 DIMENSIONAL TOOLING 5-SUBSYSTEM 80 SUBD OF WORK TOOLING AND STE

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q-1	61	3.0	455	3.048	1387	208	1595
Q-2	61						
Q-3	61		3	2.667	8	14	22
Q-4	61						
Q - 1	62						
Q-2	62						
Q-3	62						· .
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	63	6.5	1132	2.934	3321		3321
тот	AL	9.5	1590		4716	222	4938

#### TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

#### TEST/QC

4-SYSTEM8DIMENSIONAL TOOLING5-SUBSYSTEM80SUBD OF WORKTOOLING AND STE

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
	<b>F</b> 0						
Q-1	58		. <b>1</b>	2.000	2		2
Q-2	58						
0-3	58		49	5.469	268		268
Q-4	58						
Q-1	59		80	3.038	243		243
0-2	59						
Q-3	59	9.0	1526	3.080	4700		4700
Q-4	59						
Q-1	60	45.0	7821	3.259	25487		25487
Q-2	60						
0-3	60	81.0	13528	3.423	46312		46312
Q-4	60						
Q-1	61	222.0	37776	3.326	125636		125636
Q-2	61						
Q-3	61	492.0	89347	3.592	320939		320939
Q-4	61						
Q-1	62	190.5	32476	3.089	100305		100305
Q-2	62				· .		
Q-3	62	91.5	15303	3.180	48568		48668
Q-4	62						
Q-1	63	60.0	10314	3.351	34561		34561
Q-2	63						
Q-3	63	262.5	44017	3.510	154491		154491
0-4	63				· · ·	• .	
Q-1	64	76.5	12963	3.679	47694		47694
Q-2	-64		· · · ·		· .		
Q-3	64	42.0	7367	3.360	24753	· ·	24753
Q-4	64					•	
Q-1	65	7.5	1305	3.597	4694		4694
0 - 2	65				•		

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 8 DIMENSIONAL TOOLING 5-SUBSYSTEM 80 DIMENSIONAL TOOLING SUBD OF WORK TOOLING AND STE

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR Dollars	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 Q-4	65 65		39	•231	9		9
Q-1	66	1.5	283	• 253	-72		-72
тот	AL	1581.0	274195		938690		938690

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

		MAN- Months	LABOR	LABOR RATE	LABOR DOLLARS	BURDEN Dollars	LABOR + BURDEN \$	TOOL/STE Matl
Q-1 Q-2	58 58	60.0	9909	3.831	37965	25497	64462	458
Q-3 Q-4	58 58	33.0	5598	5.975	33448	10961	44409	-438
Q-1 Q-2	59 59	121.5	20832	4.481	93354	58523	151877	-20
Q-3 Q-4	59 59	471.0	82796	3.701	306427	244148	550575	89694
Q-1 Q-2	60 60	2134.5	369932	3.435	1270594	1082180	2352774	592048
Q-3 Q-4	60 60	2518.5	423083	3.398	1437438	1190420	2627858	672059
Q-1 Q-2	61 61	8863.5	1512630	3.548	5367142	4383467	9750609	3043434
Q-3 Q-4	61 61	14110.5	2558692	3.308	8462993	8159780	16622773	10266054
Q-1 Q-2	62 62	4480.5	764785	3.344	2557502	2677438	5234940	4463780
Q-3 Q-4	62 62	1782.0	299300	3.362	1006355	1241575	2247930	1878678
Q-1 Q-2	63 63	1580.5	269895	3.304	891660	979828	1871488	588332
Q-3 Q-4	63 63	294.5	49394	4.748	234520	395240	629760	380525
Q-1 Q-2	64 64	1/15.5	292626	3.420	1000717	901695	1902412	186454
Q - 4 Q - 4	64 65	127.0	- 9750	6.114	137493	170527	308120	22374
Q-2	65 65	15 0	2680	1 452	51081	42 908	2214 2214	01343
0-4	65	13+0	2707	1.000	4114	2202	0310	-14721

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$	TOOL/STE MATL
Q-1 66 Q-2 66	1.5	287	.239	-69	20	-49	3316
Q-3 66		48	8.042	386		386	35524
TOTAL	38359.0	6693543		22899126	21567509	44466635	22268688

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

		MPC	OTHER COST	SUB Total	G&A	TOTAL Cost
Q-1	58	25		64945		64945
Q-2	58					
Q = 3	58	-24		43947		43947
Q=4 0-1	20 50	_1		1 61 067		1 5 1 0 5 /
$\theta - 2$	59	-1		101000		101000
0 - 3	59	7597	6411	654277		654277
Q-4	59					021211
Q-1	60	77854	17488	3040164	57924	3098088
Q-2	<b>6</b> 0					
Q-3	60	88376	5785	3394078	64667	3458745
0-4	60					
Q-1	61	25/1/0	43741	13094954	243344	13338298
0-2	61	947401	150//5		<b>510750</b>	
Q=3	61	00/401	109400	21915113	218/24	28434532
0 - 1	62	351745	48035	10098500	160503	10268003
0-2	62	551145	40000	10000000	107705	10200000
$\overline{0}$ -3	62	148039	-24940	4249707	71331	4321038
Q-4	62					
Q-1	63	57951	625	2518396	42108	2560 504
Q-2	63					
Q-3	63	37482	-5776	1041991	17422	1059413
Q-4	63			4		
Q-1	64	19876	3702	2112444	44949	2157393
Q-2	64	01/0	27.02	3/ <b>3</b> 33 <b>7</b>	7000	2/0510
0-4	04 67	8140	2003	342231	1282	349519
0 - 1	65	18347	- 2110	1 81 705	4850	186645
0-2	65	LODAL	erto.	TOTIO	VCOT	100040
0-3	65	-2663	704	-10570	-282	-10852
Q-4	65			~ ~		

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TOTAL COST	GδA	SUB Total	OTHER COST	MPC	
4073	119	3954		687	Q-1 66 Q-2 66
39242	1147	38095		2185	Q-3 66
70179666	1243123	6893654 <b>3</b>	260953	1940267	TOTAL

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#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

#### 4-SYSTEM 8 5-SUBSYSTEM 81 CONTRACT SUPPLEMENTAL TOOLING

	TOOLING AND STE Hours Dollars	TOTAL Hours Dollars
DESIGN/ENGINEERING	622	622
LABOR AT \$ 4.286	2666	26.66
ENGR BURDEN AT \$ 5	5.272 3279	3279
TOOLING AND STE	234893	234893
LABOR AT \$ 3.398	798065	798065
TEST/QC	4593	4593
LABOR AT \$ 3.258	14964	14964
MFG BURDEN AT \$ 3	8.507 839884	839884
TOOLING/STE MATL	1275734	1275734
MPC	117271	117271
SUB-TOTAL	3051863	3051863
GEN & ADMIN	56552	56552
TOTAL COST	3108415	3108415

TIME-PHASED COST DETAIL-SEE PAGE II-648

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING

# 4-SYSTEM8CONTRACT SUPPLEMENTAL TOOLING5-SUBSYSTEM81SUBD OF WORK TOOLING AND STE

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DULL ARS	LABOR + Burden \$
Q-1	62		40	5.150	206	182	388
Q-2	62						
Q-3	62	1.5	346	1.705	590	838	1428
Q-4	62						
Q-1	63		-1	683.999	684	966	1650
Q-2	63						
Q-3	63		31	4.613	143	221	364
Q-4	63						
Q <b>-1</b>	64		22	7.091	156	150	306
Q-2	64						
Q-3	64		7	11.286	79	47	126
Q-4	64						
Q-1	65	1.5	132	4.591	606	565	1171
Q-2	65						
0-3	65		45	4.489	202	310	512
тот	AL	3.0	622	· ·	2666	3279	5945

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# TOOLING AND STE4-SYSTEM85-SUBSYSTEM81SUBD OF WORKTOOLING AND STE

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	60	40.5	6919	3.399	23519	20664	44183
Q-2	60						
Q-3	60	208.5	35139	3.229	113475	109955	223430
Q-4	60						
Q-1	61	358.5	61190	3.464	211981	185895	307976
Q-2	61					102072	571070
Q-3	61	120.0	21722	3.287	71401	67161	128562
Q-4	61					01101	100002
Q-1	62	190.5	32458	3-250	105474	146369	251023
Q-2	62			30230	103111	140347	201025
Q-3	62	153.0	25817	3.237	83577	84404	167002
Q-4	62					04400	101903
Q-1	63	60.0	10146	2.738	27770	20222	47100
Q-2	63		202.0	20100	21113	57525	0/102
Q-3	63	46.5	7813	4.709	36705	24100	70005
Q-4	63			14 10 2	20122	24100	10095
Q-1	64	51.0	8662	3.327	29017	72244	1023/1
Q-2	64		0002	JUJL	20011	12244	102101
Q-3	64	114.0	20044	3 494	70093	56363	10//0/
0-4	64		20011	34420	10005	20,243	120420
0 - 1	65	21.0	2727	5 050	10072	14761	25 ( 2 (
Q-2	65		2131		10013	10151	32024
Q-3	65	7.5	1246	5 040	4 201	5 503	33007
-			ILTU	2017	0271	2273	11884
TOT	AL	1371.0	234893		798065	839884	1637949

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC4-SYSTEM85-SUBSYSTEM81SUBD OF WORK TOOLING AND STE

		MAN- MONTHS	LABOR	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1	<b>6</b> 0	1.5	250	3.292	823		823
Q-2	60						
Q-3	60	9.0	1466	3.173	4652		4652
Q-4	60						
Q-1	61	6.0	905	2.987	2703		2703
Q-2	61						
Q-3	61	1.5	283	3.830	1084		1084
Q-4	61						
Q-1	62	3.0	447	3.349	1497		1497
Q-2	62						
Q-3	62	3.0	427	3.372	1440		1440
Q-4	62						
Q-1	63		105	3.419	359		359
Q-2	63						
Q-3	63	1.5	321	3.436	1103		1103
Q-4	63						
Q-1	64	1.5	293	3.444	1009		1009
Q-2	64						
Q-3	64		70	2.857	200		200
Q-4	64						
Q-1	65		19	3.421	65		65
Q-2	65			•			
Q-3	65		7	4.143	29		29
тот	AL	27.0	4593		14964		14964

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### 4-SYSTEM 8 5-SUBSYSTEM 81 CONTRACT SUPPLEMENTAL TODLING

		MAN- MON THS	LABOR	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	TOOL/STE MATL
Q-1	60	42.0	7169	3.395	24342	20664	45006	7017
Q-3	60 60	217.5	36605	3.227	118127	109955	228082	170992
Q-4 Q-1	60 61	364.5	62 0 9 5	3.457	214684	185895	400579	407442
Q-3	61 61	121.5	22005	3.294	72485	67161	139646	433036
Q-4 Q-1	61 62	193.5	32945	3.253	107177	146531	253708	121493
Q-3	62 62	157.5	26590	3.220	85607	85244	170851	117569
Q-4 Q-1	62 63	60.0	10250	2.812	28822	40289	69111	23664
Q-3	63 63	48.0	8165	4.659	38041	34321	72362	-28844
Q-4 Q-1	63 64	52.5	8977	3.340	29982	73494	103476	15439
Q-2 Q-3	64 64	114.0	20121	3.497	70362	56390	126752	7746
Q-4 Q-1	64 65	22.5	3888	5.027	19544	17316	36860	148
Q-2 Q-3	65 65	7.5	1298	5.025	6522	5903	12425	32
тот	AL	1401.0	240108	•	815695	843163	1658858	1275734

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 8 5-SUBSYSTEM 81 CONTRACT SUPPLEMENTAL TOOLING

	MPC	SUB Total	GεA	TOTAL COST
Q-1 60 Q-2 60	923	52946	1009	53955
Q-3 60 Q-4 60	22485	421559	8032	429591
Q = 1  61 Q = 2  61	34429	842450	15655	858105
Q-3 61 Q-4 61	36592	609274	11322	620596
Q-1 62 Q-2 62	9574	384775	6458	391233
Q-3 62 Q-4 62	9264	297684	4997	302681
Q = 1  63 Q = 2  63	2331	95106	1590	96696
Q-3 63 Q-4 63	-2841	40677	680	41357
Q = 1 64 Q = 2 64	1646	120561	2565	123126
Q-3 64 Q-4 64	2818	137316	2922	140238
Q-1 65 Q-2 65	44	3 <b>7</b> 052	989	38041
Q-3 65	6	12463	333	12796
TOTAL	117271	3051863	56552	3108415

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 8 5-SUBSYSTEM 82 HANDLING EQUIPMENT

		TCOLING	
		AND STE	TOTAL
		HOURS	HOURS
		DOLLARS	DOLLARS
DESIGN/ENGINEERIN	3	12502	12502
LABOR AT \$ 4.	380	54755	54755
ENGR BURDEN A	Г \$ 4.755	59444	59444
TOOLING AND STE		145072	145072
LABOR AT \$ 3.	401	493442	493442
TEST/QC		2135	2135
LABOR AT \$ 3.	355	7163	7163
MFG BURDEN A	r\$ 3.759	553355	553355
TOOLING/STE MATL		342074	342074
MPC		33624	33624
SUB-TOTAL		1543857	1543857
GEN & ADMIN		27983	2 <b>7</b> 983
TOTAL COST	`.	1571840	1571840

TIME-PHASED COST DETAIL-SEE PAGE II-654

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING

4-SYS	TEM		8			
5-SUB	SYS	TEM	82		HANDLING	EQUIPMENT
SUBD	OF	WORK	TOOLING	AND	STE	

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABUR DOLLAR S	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-3 61 Q-4 61	9.0	1546	4.235	6547	6050	12597
Q-1 62 Q-2 62	55.5	9563	4.368	41769	44182	85951
Q-3 62 Q-4 62	3.0	457	3.810	1741	4603	6344
Q-1 63 Q-2 63		57	6.368	363	309	672
Q-3 63	4.5	879	4.932	4335	4300	8635
TOTAL	72.0	12502		54755	59444	114199

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TOOLING AND STE4-SYSTEM85-SUBSYSTEM82SUBD OF WORKTOOLING AND STE

#### ON-SITE LABOR

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-3	59		16	4.313	69	63	132
Q-4	59		•				
Q-1	60	10.5	1770	3.419	6052	6338	12390
Q-2	60						
Q-3	60	28.5	4795	3.549	17016	15679	32695
Q-4	60						
Q-1	61	60.0	10151	3.515	35676	.34053	69729
Q-2	61						
Q-3	61	126.0	22857	3.728	85204	80262	165466
Q-4	61						
Q-1	62	190.5	32535	3.193	103888	107573	211461
Q-2	62						
Q-3	62	192.0	32275	3.254	105009	12/088	232097
Q-4	62				007/0	~~ / ~ ~	(1010
Q-1	63	58.5	10660	3.257	32 16 9	28450	61219
0-2	63		0/10	2 0 ( 0	22221	53.057	00150
Q-3	03	22.2	9419	3.008	23401	22221	02120
Q=4 01	0.5	= <b>E E</b>	0440	2 271	20040	76465	47204
0-2	04 44	22.5	9402	2.211	20747	50442	01394
0-2	64	54 0	0375	3 570	33472	47015	80487
0-4	64	JTOU	7313	5.510	33412	41017	00401
0-1	65	10.5	1764	7,272	12827	14817	27644
0-2	65	10.0	1.0.			11017	21011
0-3	65	3.0	- 593	2.715	1610	2293	3903
0-4	65	200			2020	~~~~	
Q-1	66					22	22
тот	<b>TAL</b>	844.5	145072		493442	553355	1046797

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC4-SYSTEM8HANDLING EQUIPMENT5-SUBSYSTEM82SUBD OF WORKTOOLING AND STE

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	60		6	3.000	18		18
Q-2	60						
Q-3	60		18	3.167	57		57
Q-4	60						
Q-1	61		37	3.135	116		116
ର-2	61						
Q-3	61		53	3.736	198		198
Q-4	61						
9-1	62	4.5	674	3.736	2518		2518
Q-2	62						
Q-3	62	4.5	764	2.933	2241		2241
Q-4	62		_	_			
	63		-5	3.200	-16		-16
	63		- 0 - 0				
	63	1.5	208	3.423	/12		712
0 1	03		0.01	<b>n</b> n/ <b>1</b>	1070		1.0
	04 64	1.5	321	3.301	1079		1079
Q-2	04 47		20	ວ ກາດ ກ	110		110
0-4	64		29	20 (23	110		110
0-1	65		7	3 706	22		22
$\tilde{0} - 2$	65		,	J.200	20		63
0-3	65		3	4.333	13		12
0-4	65		د.		1.2		1.7
5-1	66		20	4.700	94		94
τοτ	AL	12.0	2135		7163		7163

#### PROPRIETARY DATA OF NORTH AMERICAN ROCKWELL CORP.

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### **APRIL 1972**

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 8 5-SUBSYSTEM 82 HANDLING EQUIPMENT

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		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	TOOL/STE MATL
Q-3	59		16	4.313	69	63	132	
Q-4	59							
Q <b>-1</b>	60	10.5	1776	3.418	60 <b>7</b> 0	6338	12408	3089
Q-2	60 60	20 F	4013	2 567	17073	15679	32752	13225
3-5 0-4	60	20.0	LIO <del>F</del>	2 + 24 +	TIGID	12012	26. 120	the car have been and
Q-1	61	60.0	10188	3.513	35792	34053	698 <b>45</b>	25124
ଦ-2	61					5 ( D <b>1 5</b>	• ••• •• •• •	~ ~ ~ ~ ~ ~
Q-3	61	135.0	24456	3.760	91949	86312	178261	94233
Q = 4 G = 1	61 62	250.5	42772	3.464	148175	151755	299930	<b>5697</b> 0
0-2	62	20000	i ka 🛔 4 kan		and the second of the	· · · · · · · · · · · · · · · · · · ·		
Q-3	62	199.5	33496	3.254	108991	131691	240682	117392
Q-4	62							
Q-1	63	58.5	10112	3.275	33116	28759	61875	-3994
Q = 2 Q = 3	63	61.5	10506	3.231	33948	57557	91505	-4292
0-4	63							
Q-1	64	57.0	9783	3.274	32028	36445	68473	21269
Q-2	64	<i>[</i> <b>1 0</b>	0101			1701E	00507	10454
Q-3	64	24.0	9404	3.011	22202	41015	16609	10030
0 - 1	64 65	10.5	1771	7.256	12850	14817	27667	7077
0-2	65							
Q-3	65	3.0	596	2.723	1623	2293	3916	1025
Q-4	65							
Q <b>-1</b>	66		2.0	4.700	4	22	116	
тот	ΓAL	928.5	159709		555360	612799	1168159	342074

APRIL 1972

TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

4-SYSTEM 8 5-SUBSYSTEM 8 HANDLING EQUIPMENT

		MPC	SUB TOTAL	GεA	TOTAL COST
<b>G-</b> 3	59		132		132
Q-4	59				An and Sec.
Q <b>-1</b>	60	406	15903	303	16206
Q-2	<b>6</b> 0				
Q <b>-</b> 3	60	1739	47716	<b>9</b> 0 <b>9</b>	48625
Q-4	ാ				
Q-1	61	2123	97092	1804	93896
Q-2	61				
Q-3	61	7988	280782	5218	286000
0-4	61				
Q-1	62	4489	361389	6066	367455
0-2	62				
0-3	62	9250	367324	6166	373490
Q-4	62				
$\mathbf{O} - \mathbf{I}$	63	- 393	57488	961	58449
Q-2	63				
Q-3	63	-422	86791	1451	88242
Q-4	63				
Q <b>-1</b>	64	2267	92009	1 95 8	93967
Q-2	64				
0-3	64	3877	95130	2 0 2 4	97154
Q-4	64				
Q-1	65	2117	36851	983	37844
Q-2	65				
0-3	65	183	5124	137	5261
Q-4	65				
Q-1	66		116	3	119
тот	AL	33624	1543857	27983	157184C

OF POOR QUALITY

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#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM TOOLING

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		TOOLING AND STE HOURS	TOTAL	
		DOLLARS	DOLLARS	
DESIGN/ENGINEERING		85600	85600	
LABOR AT \$ 3.436		294104	294104	
ENGR BURDEN AT \$	4.539	388569	388569	
TOOLING AND STE		7409604	7409604	
LABOR AT \$ 3.412		25283570	25283570	
PLANN ING		572696	572696	
LABOR AT \$ 3.504		2006823	2006823	
TEST/QC		301410	301410	
LABOR AT \$ 3.422		1031533	- 1031533	
MFG BURDEN AT \$	3.044	25212209	25212209	
TOOLING/STE MATL	,	24730867	24730867	
MPC		2190524	2190524	
OTHER COST		261038	261038	
SUB-TOTAL		81399237	81399237	
GEN & ADMIN		1471025	1471025	
TOTAL COST		82870262	82870262	

TIME-PHASED COST DETAIL - SEE PAGE

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## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

### DESIGN/ENGINEERING 4-SYSTEM 8

SUBD OF WORK TOOLING AND STE

TOOLING

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58	1.5	152	4.704	715	690	1405
Q-2	58						
Q-3	58	1.5	308	3.078	948	1168	2116
0-4	58		-				
	59		3	5.333	16	· 13	29
V-2	29 50	10 5	1000				
0-4	27 50	10.5	1938	3.065	5818	6683	12501
Q = 4 $\Omega = 1$	60	19 0	2217	3 941	01/1	10115	01.05/
0-2	60	TOPA	5211	2.041	7141	12115	21250
$\hat{\mathbf{Q}} - \hat{\mathbf{Z}}$	60	27.0	4569	2,912	13303	17207	20620
Q-4	60	2	1207	20712		**	00000
Q-1	61	81.0	13931	2.794	38922	51 969	90891
0-2	61				and of a factor date.	and the set of the set	, <b>0</b> 0 7 <b>1</b>
Q-3	61	102.0	18528	3.157	58492	77496	135988
Q-4	61						
Q-1	62	105.0	17939	3.913	70191	82749	152940
Q-2	62	•					
Q-3	62	58.5	9926	3.315	32904	52308	85212
Q-4	62						
Q-1	63	28.5	4893	4.253	20808	26546	47354
Q-2	63	<b>24 F</b>			بدو سر و بن ود.		
0-4	63	34.5	5742	4.4.33	25455	30832	56287
0-1	66	10 5	2202	2 072	12100	22004	26004
0-2	64	1.702	201.0	2012	12100	22904	50064
ã-3	64	4.5	778	3, 284	2555	2521	6086
0-4	64		, , , ,		the the the	± 2.5 €.	0000
Q-1	65	1.5	256	4.969	1272	1493	2765
Q-2	65						
Q-3	65,		78	5.949	464	665	1129
тот	AL	493.5	85600	·	294104	388569	682673

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TOOLING

	TUOLING	AND	STE	
4-SYSTEM	8			
SUBD OF WORK	TOOLING	AND	STE	

## ON-SITE LABOR

		MAN-	LABOR	LABOR	LABOR	<b>BURDEN</b>	LABOR +
		MUNIHS -	HUUKS	RAIE	DULLARS	DOLLARS	BURDEN \$
Q-1	58	58.5	9756	3.818	37248	25807	63055
Q-2	58						
Q-3	58	31.5	5241	6.150	32232	9793	42025
Q-4	58						
Q-1	59	121.5	20749	4.487	93095	58510	151605
Q-2	59						
Q-3	59	451.5	79388	3.728	295978	237528	533506
Q-4	59						
Q-1	60	2143.5	371556	3.439	1277811	1109669	2387480
Q-2	60						
Q-3	60	2728.5	458474	3.375	1547305	1340717	2888022
0-4	60						
Q-1	61	9139.5	1559691	3.546	5530214	4531297	10061511
Q-2	61						
Q-3	61	14541.0	2636675	3.320	8753065	8725373	17478438
Q-4	61						
Q-1	62	5608.5	957201	3.327	3184607	3485648	6670255
Q-2	62						
Q-3	62	2380.5	400027	3.306	1322453	1669037	2991490
Q-4	62						
Q-1	63	1738.5	296731	3.376	1001830	1133150	2134980
Q-2	63						
Q-3	63	653.5	109761	3.586	393617	690 <b>7</b> 28	1084345
Q-4	63						
Q-1	64	2103.0	358874	3.933	1411372	1527398	2938770
Q-2	64						
Q-3	64	687.0	120997	2.385	288613	362423	651036
Q-4	64						
Q-1	65	105.0	18170	5.255	95480	100260	195740
Q-2	65						
Q-3	65	37.5	6261	2.917	18261	19054	37315

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## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	TOOLING	AND	STE	
4-SYSTEM	8			TOOLING
SUBD OF WORK	TOOLING	AND	STE	

		MAN- MONTHS	LABOR HOURS	LABOR	LABOR DOLLARS	BURDEN Doll Ars	LABOR + BURDEN \$
0-4 6	55						
Q-1 6	6		4	.750	3	42	45
Q-2 6	6						
Q-3 6	6		48	8.042	386		386
τοτα	L	<b>42529.</b> 0	7409604		25283570	25026434	50310004

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

		PLANNING	3		BOOT THE
4-SYSTE	M	8			TOOLING
SUBD OF	WORK	TOOLING	AND	STE	

•		MAN-	LABOR	LABOR	LABOR	<b>BUR DEN</b>	LABOR +
		MONTHS	HOURS	RATE	DOLLARS	DOLL ARS	BURDEN \$
Q-1	61	715.5	122184	3+174	387789	49103	436892
Q-2	61						
Q-3	61	343.5	62214	3.069	190907	26371	217278
Q-4	61						
Q-1	62		7	•142	-1		-1
Q-2	62						
Q-3	62						
Q-4	62						·
0-1	63						
Q-2	63						
Q-3	63	2236.5	375698	3.680	1382427	64642	1447069
Q-4	63			-			
Q-1	64	36.0	6021	2.550	15354	16464	31818
Q-2	64						
Q-3	64	10.5	1963	7.200	14134	13451	27585
Q-4	64						
Q-1	65	19.5	3351	3.297	11047	9138	20185
Q-2	65						
Q-3	65	6.0	1123	4.185	4700	6016	10716
Q-4	65						
Q-1	66		3				
Q-2	66						
Q-3	<b>6</b> 6	1.5	132	3.530	466	590	1056
тот	TAL	3369.0	572696		2006823	185775	2192598

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 8

TOOLING

SUBD OF WORK TOOLING AND STE

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58		1	2.000	2		2
Q-2	53						
Q-3	58		49	5.469	268		268
Q-4	58						
Q-1	59		80	3.038	243		243
0-2	59						
Q-3	59	9.0	1526	3.080	4700		4700
Q - 4	59						
Q-1	60	46.5	8031	3.259	26340		26340
0-2	60						
Q-3	60	90.0	15231	3.397	51745		51745
Q-4	60						
Q-1	61	229.5	39176	3.317	129950		129950
Q-2	61						•
Q-3	61	510.0	92496	3.581	331205		331205
0-4	61						
Q-1	62	228.0	38973	3.135	122162		122162
Q-2	62						
0-3	62	112.5	18980	3.207	60861		60861
Q-4	62						
Q-1	63	69.0	11770	3.358	39523		39523
0-S	63						
Q-3	63	283.5	47607	3.509	167072		167072
()-4	63						
Q-1	64	97.5	16532	3.677	60795		60795
Q-2	64						
6-3	64	49.5	8785	3.432	30147		30147
0-4	64				–		
<u>u-1</u>	65 ( 5	10.5	1//1	3.640	6447		6447
<b>u</b> -2	65				<b>_</b> -		
U-3	65		49	1.041	51		51

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	TEST/QC		TOOLING
4-SYSTEM	8		
SUBD OF WORK	TOOLING AN	D STE	

	MAN- MON TH S	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-4 65 Q-1 66	1.5	303	• 73	22		22
TOTAL	1737.0	301410		1031533		1031533

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 8 TOOLING SUBD OF WORK TOOLING AND STE

		MAN- MONTHS	LABOR HOURS	LABOR Rate	LABOR Dollars	BUR DEN Doll Ars	LABOR + Burden \$	TOOL/STE MATL
Q-1	58	60.0	9909	3.831	37965	26497	64462	458
Q-2 Q-3	58 58 59	33.0	5598	5.975	33448	10961	44409	-438
0-4 0-1 0-2	23 59 59	121.5	20832	4.481	93354	58523	151877	-20
0-3 0-4	59 59	471.0	82812	3.701	306496	244211	550707	8969 <b>4</b>
Q-1 Q-2	60 60	2208.0	382854	3.430	1313292	1121784	2435076	603602
0-3 0-4	60 60	2845.5	478274	3.371	1612353	1358044	2970397	861421
Q-1 Q-2	61 61	10165.5	1734982	3.508	6086875	4632369	10719244	3487324
0-3 0-4	61 61	15496.5	2809913	3.322	9333669	8829240	18162909	10927409
Q-1 Q-2	62 62	5941.5	1014120	3.330	3376959	3568397	6945356	4801644
Q-3 0-4	62 62	2551.5	428933	3.302	1416218	1721345	3137563	2253043
Q = 1 Q = 2	63 63	1836.0	313394	3.389	1062161	1159696	2221857	672399
Q-3 Q-4	63 63	3208.0	538808	3.654	1968571	786202	2754773	406708
0 - 1 0 - 2	64 64	2256.0	384809	3.900	1500621	1566846	3067467	395234
Q-3 0-4	64 64	751.5	132523	2.531	335449	379405	714854	115946
Q-1 Q-2	65 65	136.5	23548	4.852	114246	110891	225137	83678
0-3 0-4	65 65	43.5	7511	3.126	23476	25735	49211	-6550
Q-1	66	1.5	310	. 81	25	42	67	3791

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TOOLING

4-SYSTEM 8 SUBD OF WORK TOOLING AND STE

	MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN DOLL ARS	LABOR + Burden \$	TOOL/STE MATL
Q-2 66							
Q-3 66	1.5	180	4.733	852	590	1442	35524
TOTAL	48128.5	8369310		28616030	25600778	54216808	2473086 <b>7</b>

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

### TOOLING

4-SYSTEM 8 SUBD OF WORK TOOLING AND STE

			OTHER	SUB		TOTAL
		MPC	COST	TOTAL	GEA	COST
Q-1	58	25		64945		64945
Q-2	58					01212
Q-3	58	-24		43947		43947
0-4	58					
Q-1	59	-1		151856		151856
Q-2	59					
Q-3	59	7597	6411	654409		654409
Q-4	59					
Q-1	60	79373	17573	3135624	59743	3195367
Q-2	60					
Q-3	60	113277	5785	3950880	75276	4026156
Q-4	60					1 1/ 100 100 UK 100 UK
Q-1	61	294679	43741	14544588	270289	14815277
Q-2	51					
Q-3	61	923366	159465	30173149	560707	30733856
0 - 4	61					
$\hat{\mathbf{n}} - \hat{1}$	62	378369	48035	12173404	204329	12377733
$\tilde{0} - 2$	62	210205	مين في ميا عين <u>و</u>	12119107	34 m % - 9 , 4 4 4 4	12011100
0-3	62	177538	-24940	5543204	03043	5636947
0-4	62	11, 550	21710	2242204	12042	2020241
0-1	62	65653	625	2060 524	40500	2010024
0-2	62		U C D	2300334	49500	2010024
0-3	63	30528	-5776	2105222	52474	2249457
0-4	63	39320	-9110	3137233	22424	. 2240001
0-1	64	10100	2702	2 E A O E 2 E	74/54	2502100
Q = 1	61	46106	5102	2202222	14024	2202T9A
Q-2	04	10100	2602	074505	10/20	A
	04	42182	2003	610080	18052	842231
Q-4	04	05007				
Q-1	02	25027	2110	335952	8963	344915
Q-2	65		<b>.</b>			
Q-3	05	-1168	704	42197	1127	43324
⊌-4	65	1000 c				
Q-1	66	786		4644	139	4783

C-8.

TOTAL COST

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM			8			TOOLING			
SUBD	OF	WORK	TOOLING	AND	STE				
			OTI	HER .		SUB			
		MPO	C CI	DS T	TO	TAL	G	3	A

			·		
Q-2 66					
Q-3 66	2185		39151	1179	40330
TOTAL	2190524	261038	81399237	1471025	82870262

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 8 TOCLING

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$	ENGR MATL
0-1	58	60.0	9909	3.831	37965	26497	64462	
Q-2	53 50				· · · · · · ·			
0-4	28 59	33.0	5598	5.975	3344.8	10961	44409	
$0^{-1}$	59 50	121.5	20832	4.481	93354	58523	151877	
Q-3	59 59	471.0	82812	3.701	306496	244211	550707	
0-1	<b>6</b> 0	2208.0	382854	3.430	1313292	1121784	2435076	
9 - 3	60 60	2845.5	478274	3.371	1612353	1358044	2970397	
(-4) (-1) (-2)	00 61 61	10165.5	1734982	3.508	6086875	4632369	10719244	
Q-3 0-4	61 61	15496.5	2809913	3.322	9333669	8829240	18162909	
Q = 4 Q = 1 Q = 2	62 62	5941.5	1014120	3.330	3376959	3568397	6945356	
Q-3	62 62	2551.5	428933	3.302	1416218	1721345	3137563	
Q = 1 Q = 1	63 63	1836.0	313394	3.389	1062161	1159696	2221857	
Q = 3	63 63	3208.0	538808	3.654	1968571	786202	2754773	
0 - 1 0 - 2	64 64	2256.0	384809	3.900	1500621	1565846	3067467	
Q-3 0-4	64 64	751.5	132523	2.531	33544 9	379405	714854	
$\overline{Q} - 1$ $\overline{Q} - 2$	65 65	136.5	23548	4.852	114246	110891	225137	
Q-3 0-4	65 65	43.5	7511	3.126	23476	25 <b>7</b> 35	49211	
0-1	66	1.5	310	. 81	25	42	67	

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 8 TOOLING

	MAN- MONTHS	LABOR HOURS	LABOR Rate	LABOR Dollars	BUR DEN DOLL ARS	LABOR + Burden \$	ENG MAT
Q-2 66	15	190	4 722	057	500	144.2	
00 2-9	1.00	100	4.122	072		1442	
TOTAL	48128.5	8369310		28616030	25600778	54216808	

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# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 8 TOOLING

		TOOL/STE MATL	TOTAL MATERIAL	MPC	OTHER COST	SUB Total	G & A	TOTAL Cost
0 - 1	58	<b>4</b> 58	458	25		64945		64945
Q-3	58 58	-438	-438	-24	i	43947	i na stario da stario	<b>4</b> 394 <b>7</b>
Q-4 Q-1	58 59	-20	-20	-1		151856		151856
Q-2 Q-3	59 59	89694	85694	, 7597	6411	654409		654409
Q-4 0-1	59 60	603602	603602	79373	17573	3135624	59743	3195367
Q-2 Q-3	60 60	861421	861421	113277	5785	3950880	75276	4026156
Q-4 Q-1	60 61	3487324	3487324	294679	43741	14544988	270289	14815277
0-2 0-3 0-4	01 61 61	10927409	10927409	923366	159465	30173149	560707	30733856
0 - 1 0 - 2	62 40	4801644	4801644	378369	48035	12173404	204329	12377733
Q-2 Q-3	62 62	2253043	2253043	177538	-24940	5543204	93043	5636247
Q-1 Q-2	63 63	672399	672399	65653	625	2960534	49500	3010034
Q-3	63 63 43	405708	406708	39528	-5776	3195233	53424	3248657
Q-1	64 64	395234	395234	42132	3702	3508535	74654	3583189
Q-3 0-4	64 64	115946	115946	42182	3603	876585	18652	89523 <b>7</b>
Q = 1 Q = 1 Q = 2	65 65	83678	83678	2502 <b>7</b>	2110	335952	8963	344915
Q-3	65 65	-6550	-6550	-1168	704	42197	1127	43324
0-1	66	3791	3791	786		4644	139	4783

OF POOR QUALITY

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# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# 4-SYSTEM 8 TOOLING

.

TOTAL COST	G & A	SUB TOTAL	OTHER COST	MPC	TOTAL MATERIAL	TOOL/STE MATL	
4033(	1179	39151		2185	35524	35524	Q-2 66 Q-3 66
82870262	1471025	81399237	261 03 8	2190524	24730867	24730867	TOTAL





### OTHER PROGRAM ELEMENTS WBS 9.0

This item has been established to collect those costs which can not be associated with any other element of the Work Breakdown Structure. The level 5 items and their definitions contained in Other Program Elements are:

			Cost	Data
9.90 9.91 9.92 9.93 9.94 9.95 9.96 9.97 9.98	Preproduction Training Publications Photographic Photo Lab Reliability Producibility Computing Photo-Template Lab Termination Costs	<pre>\$ 1,815,287 \$ 868,720 \$ 2,310,296 \$ 442,260 \$ 825,886 \$ 0 \$ 1,948,323 \$ 255,698 \$ 7.574.353</pre>	pg. pg. pg. pg. pg. pg.	II-680 II-707 II-713 II-720 II-724 II-728 II-735 II-739
	Total WBS 9.0	\$ 16,040,823	pg.	II-746

### 9.90 Preproduction

Costs contained in this item represent the labor and purchased services associated with the design of initial layout, set-up and rearrangement of the production department including the assembly and test areas for the B-70 program. These costs include installation of assembly jigs and fixtures, utilities where required for operation of contract tools, material handling equipment, storage equipment, and work benches. Also included is the effort to determine the types of special tools and special test equipment required to support assembly operations. Excluded from this item is the cost incurred in connection with the design, acquisition and installation of contract instrumentation and test equipment and the installation of simulators and related equipment.

#### 9.91 Training

This item contains all identifiable engineering labor, burden and material costs associated with the personnel training program and the supplying of technical services to the customer.

### 9.92 Publications

Costs reported in this item represent all identiable Engineering and Logistics labor, burden and material charges required to prepare and deliver publications, charts, slides, films etc. as required by the contract. Included are the publication costs associated with the preparation and delivery of:

- a) Flight Manuals
- b) Maintenance Instructions Manuals
- c) Structural Repair Manuals
- d) Illustrated Parts Breakdown Manual
- e) Parts Application Data List

11-675

SD72-SH-0003

\$ 1,815,287

868.720

\$ 2,310,296

\$

II**-**676

SD72-SH-0003

## f) Numerical Drawing List

- g) Modification Instruction Manual
- h) Technical Training Transparencies and Slides
- i) Motion Pictures

Costs to research, analysis and prepare the technical data included in these items is not contained in this WBS item.

### 9.93 Photographic Photo Lab

Costs identified to this WBS item included the expenditures for developing, processing, and analyzing photographic data generated by test instrumentation.

### 9.94 Reliability

Contained in this WBS item are the identifiable costs for the establishment and implementation of an organized reliability and crew safety program. This program assured retention of reliability through design, development, testing, manufacturing and flight testing.

### 9.95 Producibility

No costs were assigned to this item as the data could not be identified from the subsystems.

### 9.96 Computing

This WBS item contains the technical programming effort related to the problems solving and data analysis functions identified specifically as being B-70 unique. Machine costs and general programming effort not related specifically to the B-70 program are not included in this item but are contained in the "Other Cost Dollars" Element of Cost.

9.97 Photo-Template Lab

This WBS item contains all identifiable costs associated with the fabrication of instrument panels utilizing the photo-template processes.

### 9.98 Termination Costs

Data contained in this WBS item includes the identifiable in-house and subcontractor costs paid as the result of the termination of a contract for the convenience of the government. These costs are associated primarily with the termination of Air Vehicle No. 3 and the program redirection from 11 YB-70's to 3XB-70's. Termination costs associated with the initial program redirection from the RS-70 program to the XB-70 program can not be identified. They are included in the in-house and supplier costs displayed in the subsystems. Included are such items as severance pay, settlement of suppliers' claims, preparation of redirection documents, disposal of redirected materials, processing of redirected employees, etc.

# WBS CODE: 9.0

825,886

442,260

\$

\$

\$

\$

\$ 255.698

\$ 7,574,353

0

1,948,323

Space Division North American Rockwell

## COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYS TEM 9 OTHER PROGRAM ELEMENTS

		5-SUBSYS 90 HOURS DOLLARS	5-SUBSYS 91 Hours Dollars	5-SUBSYS 92 HOURS DOLLARS	5-SUBSYS 93 HOURS DOLLARS
DESIGN/ENGINEERING		8922	104274	210947	45644
ENGR BURDEN AT \$	4.682	39181	401368	1105284	204502
PRODUCTION		64772			
LABOR AT \$ 3.110 SHOP SUPPORT		201412			
LABOR AT \$ 3.022		22321			•
TOOLING AND STE		100268			
LABOR AT \$ 3.508		351743			
PLANNING		960			
LADUK AT \$ 5.205		12562	· ·		
$1\Delta B \cap R  \Delta T  \$  3_{-}475$		43657			
MFG BURDEN AT \$	3.650	678796			
ENGR MATERIAL		11413	1434	129711	
MFG MATERIAL		65232			
TOOLING/STE MATL		289149			
MPC		3/169	1.36	21244	
UTHER CUST		4 9 4	903		
SUB-TOTAL		1781730	866410	2263529	435092
GEN & ADMIN		33557	2310	46767	7168
TOTAL COST		1815287	868720	2310296	442260

SUBDIVISION OF WORK COST DETAIL - SEE PAGE II-680

II-707 II-713 II-720

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### COST BREAKDOWNS B-70 AIRCRAFT STUDY

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4-SYSTEM 9 OTHER PROGRAM ELEMENTS

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		•	5-SUBSYS 94 HOURS DOLLARS	5–SUBSYS 96 HOURS DOLLARS	5-SUBSYS 97 HOURS DOLLARS	5-SUBSYS 98 HOURS DOLLARS
DESIGN/ENGINE LABOR AT \$ ENGR BURDEN	ERING 4.814 AT \$ 4.0	682	83785 418454 393933	194434 1005464 911337	28164 136957 115198	
PRODUCTION LABOR AT \$ SHOP SUPPORT	3.110					
TOOLING AND S	3.022 TE 3.508					
PLANNING LABOR AT \$	3.203					
TEST/QC LABOR AT \$	3.475					
MFG BURDEN	AT \$ 3.0	550				
ENGR MATERIAL MFG MATERIAL TOOLING/STE M MPC	ATL					
OTHER COST						7414547
SUB-TOTAL			812387	1916801	252155	7414547
GEN & ADMIN			13499	31522	3543	159806
TOTAL COST			825386	1948323	255698	7574353
	SUBDIVISION O COST DETAIL -	F WORK SEE PAGE	II-724	II <b>-7</b> 28	II <b>-</b> 735	II <b>-7</b> 39

# COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 9 OTHER PROGRAM ELEMENTS

TOTAL HOURS DOLLARS DESIGN/ENGINEERING 677170 LABOR AT \$ 4.814 3260198 ENGR BURDEN AT \$ 4.682 3170803 PRODUCT ION 64772 LABOR AT \$ 3.110 201412 SHOP SUPPORT 7386 LABOR AT \$ 3.022 22321 TOOLING AND STE 100268 LABOR AT \$ 3.508 351743 **PLANN ING** 960 -LABOR AT \$ 3.203 3075 TEST/QC 12562 LABOR AT \$ 3.475 43657 MFG BURDEN AT \$ 3.650 678796 ENGR MATERIAL 142558 MFG MATERIAL 65232 TOOLING/STE MATL 289149 MPC 58549 OTHER COST 7455158 SUB-TOTAL 15742651 GEN & ADMIN 298172 -----TOTAL COST 16040823

SUBDIVISION	<b>O</b> I	WOF	IK	
COST DETATL	-	SEE	PAGE	TT-746

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## COST BREAKDOWNS B-70 AIRCRAFT STUDY

## 4-SYSTEM 9 5-SUBSYSTEM 90 PREPRODUCTION

		DESIGN /ENGR HOURS DOLLARS	PROD Hours Dollars	TOOLING AND STE HOURS DOLLARS	TOTAL Hours Dollars
DESIGN/ENGINEERING		<b>39</b> 00	4431	591	8922
LABOR AT \$ 4.273		18816	16668	2644	38128
ENGR BURDEN AT \$	4.392	14745	21142	3294	39181
PRODUCTION			64772		64772
LABOR AT \$ 3.110			201412		201412
SHOP SUPPORT		7386			7386
LABOR AT \$ 3.022		22321			22321
TOOLING AND STE				100268	100268
LABOR AT \$ 3.508				351743	351743
PLANN ING			562	398	960
LABOR AT \$ 3.203			1826	1249	3075
TEST/QC		653	10639	1270	12562
LABOR AT \$ 3.475		2005	37275	4377	43657
MEG BURDEN AT \$ 3	3.650	25902	286833	366061	678796
ENGR MATERIAL		11413			11413
MFG MATERIAL			65232		65232
TOOLING/STE MATL				289149	289149
MPC		1438	6781	28950	37169
OTHER COST		163	291		454
SUB-TOTAL		96803	637460	1047467	1781730
GEN & ADMIN		1837	11951	19769	33557
TOTAL COST		98640	649411	1067236	1815287

TIME-PHASED COST				
DETAIL-SEE PAGE	II <b>-</b> 681	II-686	II <b>-</b> 692	II <b>-</b> 698

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING

4-SYSTEM	9	
5-SUB SYSTEM	90	PREPRODUCTION
SUBD OF WORK	DESIGN/ENGINEER]	NG

		MAN-	LABOR	LABOR	LABOR	<b>BUR DEN</b>	LABOR +
		MONTHS	HOURS	RATE	DOLLARS	DOLL ARS	BURDEN \$
Q-1	60	7.5	1303	5.232	6817	4899	11716
Q-2	60						
Q-3	60	10.5	1862	4.523	8422	6 948	15370
Q-4	60						
Q-1	61	3.0	525	5.105	2680	1781	4461
Q-2	61						
Q-3	61	1.5	185	4.081	755	995	1750
Q-4	61						
Q-1	62		25	5.680	142	115	257
Q-2	62						
Q-3	62					7	7
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	63						
						· · · · ·	
TOT	<b>AL</b>	22.5	3900		18816	14745	33561

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	SHOP SUPPORT	
4-SYSTEM	9	
5-SUB SYSTEM	90	PREPRODUCTION
SUBD OF WORK	DESIGN/ENGINEERI	NG

		MAN- MON THS	LABOR HOURS	LABOR RATE	L ABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	60	12.0	2124	2.803	5953	8679	14632
Q-2	60						
Q-3	60	22.5	3793	3.079	11680	12042	23722
Q-4	60						
Q-1	61	9.0	1435	3.203	4596	4777	9373
Q-2	61						
Q-3	61		34	2.706	92	404	496
Q-4	61						
Q-1	62						
Q-2	62						
Q-3	62						
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	63						
тот	TAL	43.5	7386		22321	25902	48223

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## TEST/QC

4-SYSTEM9PREPRODUCTION5-SUBSYSTEM90SUBDOFWORKDESIGN/ENGINEERING

	MAN-	LABOR	LABOR	LABOR	<b>BUR DEN</b>	LABOR +
	MONTHS	HOURS	RATE	DOLLARS	DOLLARS	BURDEN \$
<b>Q-1</b> 60	1.5	249	3.446	858		858
Q-2 60	1					
Q-3 60	1.5	275	2.964	815		815
Q-4 60	) .			~		
Q-1 61		110	2.545	280		280
Q-2 61						
Q-3 61		19	2.737	52		52
Q-4 61						
Q-1 62						
Q-2 62						
Q-3 62						
Q-4 62						
Q-1 63	i i i i i i i i i i i i i i i i i i i					
Q-2 63	,					
Q-3 63	i					
TOTAL	3.0	653		2005		2005

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM9PREPRODUCTION5-SUBSYSTEM90SUBD OF WORK DESIGN/ENGINEERING

		MAN- MONTHS	LABOR	LABOR RATE	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$	ENGR MATL
Q-1	60	21.0	3676	3.707	13628	13578	27206	2596
Q-2	60							
Q-3	<b>6</b> 0	34.5	5930	3.527	20917	18990	39907	7497
Q-4	60							
Q-1	61	12.0	2070	3.650	7556	6558	14114	1295
Q-2	61							
Q-3	61	1.5	238	3.777	899	1 399	2298	8
Q-4	61						,	
0-1	62		25	5.680	142	115	257	17
Q-2	62							
Q-3	62					7	7	
Q-4	62							
Q-1	63							
Q-2	63							
Q-3	63							
тот	AL	69.0	11939		43142	40.647	83789	11413

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM9PREPRODUCTION5-SUBSYSTEM90SUBD OF WORKDESIGN/ENGINEERING

		MPC	COST	SUB Total	GEA	COST
Q-1	60	341		30143	574	30717
Q-2 Q-3	60 60	986	453	48843	931	49774
Q-4	60				<u>-</u>	
Q-1	61	109	29	15547	289	15836
Q-2	61					
Q-3	61	1	5	2312	43	2355
Q-4	61					
Q-1	62	1		275	5	280
Q-2	62					
Q-3	62	x		7		7
Q-4	62					
Q-1	63		-485	-485	-8	-493
Q-2	63				•	
Q-3	63		161	161	3	164
тот	AL	1438	163	96803	1837	98640

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## DESIGN/ENGINEERING 4-SYSTEM 9 5-SUBSYSTEM 90 PREPRODUCTION SUBD OF WORK PRODUCTION

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	60		8	2.625	21	28	49
Q-2	60						
Q-3	60	4.5	751	3.073	2308	2810	5118
Q-4	<b>6</b> 0						
Q-1	61	12.0	2019	3.340	6743	7835	14578
Q-2	61						
Q-3	61	1.5	254	3.839	975	1906	2881
Q-4	61						
Q-1	62	1.5	161	3.472	559	744	1303
Q-2	52						
Q-3	62					40	40
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	63		72	8.347	601	404	1005
Q-4	63						
Q-1	64	7.5	1165	4.688	5461	7031	12492
Q-2	64						
Q-3	64		1			344	344
тот	AL	27.0	4431		16668	21142	37810

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PREPRODUCTION

	PRODUCTION
4-SYSTEM	9
5-SUB SYSTEM	90
SUBD OF WORK	PRODUCTION

		MAN- MON TH S	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	60	15.0	2535	3.409	8641	11495	20136
Q-2	60						
Q-3	60	147.0	24680	3.075	75894	100779	176673
Q-4	60						
Q-1	61	129.0	21979	3.123	68643	89110	157753
Q-2	61						
Q-3	61	46.5	8500	2.963	25188	47942	73130
Q-4	61						
Q <b>-1</b>	62	18.0	3062	2.910	8911	17899	26810
Q-2	62						
Q-3	62					594	594
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	63	9.0	1587	4.100	6506	6180	12686
Q-4	63						
Q-1	64	15.0	2502	3.133	7839	13077	20916
Q-2	64						
Q-3	64		-73	2.877	-210	-415	-625
TOT	TAL	379.5	64772		201412	286661	488073

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PLANNING 4-SYSTEM 9 5-SUBSYSTEM 90 SUBD OF WORK PRODUCTION

PREPRODUCTION

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$
Q-3 61 Q-4 61		40	2.875	115	18	133
Q-1 62 Q-2 62		Ģ	3.556	32	6	38
Q-3 62 Q-4 62					. <b>1</b>	1
Q-1 63 Q-2 63						.*
Q-3 63 Q-4 63	3.0	483	3.255	1572		1572
Q-1 64 Q-2 64		30	3.500	105	145	250
Q-3 64				2	2	4
TOTAL	3.0	562		1826	172	1998

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TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	TEST/QC	
4-SYSTEM	9	PREPRODUCTION
5-SUBSYSTEM	90	
SUBD OF WORK	PRODUCTION	

# ON-SITE LABOR

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	60	1.5	378	2.915	1102		1102
Q-2 Q-3	60 60	13.5	2300	3.499	8048		8048
Q-4 Q-1	60 61	19.5	3450	3.171	10939		10939
Q-2 0-3	61 61	13.5	2579	3, 300	8767		8767
Q-4	61		1/7/	4 100	6000	. ,	(000
Q-2	62 62	0.eC	1020	4.188	6809		6803
Q-3 Q-4	62 62				• .		
Q-1 Q-2	63 63						
Q-3 0-4	63 63		13	7.462	97		97
Q-1	64 64	1.5	284	5.229	1485		1485
Q-3	64 64		9	3.111	2.8		28
тот	<b>AL</b>	58.5	10639		37275		37275

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PREPRODUCTION

4-SYSTEM 9 5-SUBSYSTEM 90 SUBD OF WORK PRODUCTION

		MAN- MONTHS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	MFG Matl
Q-1	60	16.5	2921	3.343	9764	11523	21287	585
Q-2	60							
Q-3	60	165.0	27731	3.110	86250	103589	189839	23519
Q-4	60							
Q-1	61	160.5	27448	3.145	86325	96 945	183270	22200
Q-2	61							
Q-3	61	61.5	11373	3.081	35045	49866	84911	5808
0-4	61							
Q-1	62	28.5	4858	3.358	16311	18649	34960	1933
Q-2	62							
Q-3	62					635	635	1516
0-4	62							
Q-1	63							
Q-2	63		×					
Q-3	63	12.0	2155	4.072	8776	6584	15360	6871
Q-4	63							
Q-1	64	24.0	3981	3.740	14890	20253	35143	2778
Q-2	64							
Q-3	64		-63	2.857	-180	-69	-249	22
тот	AL	468.0	80404		257181	307975	565156	65232

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PREPRODUCTION

4-SYSTEM 9 5-SUBSYSTEM 90 SUBD OF WORK PRODUCTION

	MPC	OTHER COST	SUB TOTAL	G & A	TOTAL COST
Q-1 60	77		21949	418	22367
Q-2 60 Q-3 60	3093		216451	4124	220575
Q-4 60 Q-1 61	1876		207346	3853	211199
Q-2 61 Q-3 61	491		91210	1695	92905
Q=4 61 Q=1 62 Q=2 62	152		37045	622	37667
Q-3 62	119		2270	38	2308
Q = 1  63 Q = 2  63					
Q-3 63 Q-4 63	677	291	23199	388	23587
Q-1 64 Q-2 64	296		38217	813	39030
Q-3 64			-227		-227
TOTAL	6781	291	637460	11951	649411

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESIGN/ENGINEERING						
4-SYSTEM	9						
5-SUB SYSTEM	90	PREPRODUCTION					
SUBD OF WORK	TOOLING AND S	TE					

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	60		19	4.105	78	71	149
Q-4	60						
Q-1	61		5	3.200	16	17	33
Q-2	61						
Q-3	61	1.5	195	3.621	706	865	1571
Q-4	61						
0-1	62						
0-2	62						
Q-3	62						
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	63		30	3.400	102	166	268
Q-4	63						
Q-1	64	1.5	175	4.543	795	1062	1857
Q-2	64						
Q-3	64	1.5	. 167	5.671	947	1113	2060
тот	TAL.	4.5	591		2644	3 2 9 4	5938

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	TOOLING	AND	STE	
4-SYSTEM	9			DEFENDING
5-SUB SYSTEM	90			FVELODOCITON
SUBD OF WORK	TOOLING	AND	STE	

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	59		5	2.200	11	13	24
0-4	59						
Q-1	60	9.0	1610	3.376	5436	4612	10048
Q-2	60						
Q-3	60	145.5	24383	3.309	80687	79804	160491
Q-4	60						
Q-1	61	94.5	16098	3.262	52518	-53803	106321
Q-2	61						
Q-3	61	70.5	12904	3.860	49810	48540	98350
0-4	61						
Q-1	62	39.0	6652	3.411	22691	25061	47752
Q-2	62						
Q-3	62	9.0	1464	2.794	4091	7060	11151
Q-4	62						
Q-1	63	18.0	3083	3.299	10171	11746	21917
Q-2	63						
Q-3	63	66.0	11007	3.607	39698	40 325	80023
Q-4	63						
Q-1	64	96.0	16320	3.968	64757	67892	132649
Q-2	64						
0-3	64	31.5	5422	3.186	17272	20190	37462
Q-4	64						
Q-1	65	7.5	1260	3.439	4333	6509	10842
Q-2	65						
Q-3	65		60	4.467	268	297	565
тот	TAL	586.5	100268		351743	365852	717595

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# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	PLANNING	
4-SYS TEM	9	PREPRODUCTION
5-SUB SYSTEM	90	
SUBD OF WORK	TOOLING AND ST	ſE

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	60		8	2.625	21		21
Q-4	60						
Q-1	61		98	2.888	283		283
Q-2	61						
Q-3	61						
Q-4	61				·		
Q-1	62						
Q-2	62						
Q-3	62						
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	63	1.5	262	3.179	833	97	930
Q-4	63						
Q-1	64		26	3.615	94	94	188
Q-2	64						
Q-3	64		4	4.500	18	18	36
тот	AL	1.5	398		1249	209	1458

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# TEST/QC 4-SYSTEM 9 PREPRODUCTION 5-SUBSYSTEM 90 SUBD OF WORK TOOLING AND STE

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-1 Q-2	60 60		19	3.105	59		59
Q-3 Q-4	60 60	1.5	296	3.247	961		961
Q-1 Q-2	61 61	3.0	476	3.431	1633		1633
Q-3 Q-4 Q-1	61 61 62		37	2.919	108		108
Q-2 Q-3	62 62		4	3 250	1 2		12
Q-4 Q-1	62 63		т 5	4.000	20		10
$\bar{0}-2$ 0-3	63 63	1.5	227	3.573	2.U 81.1		20
Q-4 0-1	63 64		50	3 909	230		220
Q-2	64 64		<u> </u>	2 005	154		250
Q-4	64 65	· .	106	2 4 4 2	100		120
TOT	Δ1	6-0	1270	3.042	00C 4277		00C
		0.00	LCIU		7211		4511
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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM9PREPRODUCTION5-SUBSYSTEM90SUBDSUBDCFWORKTOOLINGANDSTE

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$	TOOL/STE MATL
Q-3	59		5	2.200	11	13	24	
Q-4	59			:			x	
Q-1	<b>6</b> 0	9.0	1629	3.373	5495	4612	10107	. 405
Q-2	60							
Q-3	60	147.0	24706	3.309	81747	79875	161622	42374
Q-4	60	<u> </u>						
	61	97.5	16677	3.265	54450	53820	108270	35983
Q = 2	61	7.0						
Q-3	61	12.0	13136	3.854	50624	49405	100029	72363
Q-4	61							
Q-1	62	39.0	6552	3.411	22691	25061	47752	56015
0-2	62							
Q-3	62	9.0	1468	2.796	4104.	7060	11164	41095
Q-4	62	10.0	2000	<u> </u>				
<u>u-1</u>	63	18.0	3088	3.300	10101	11746	21937	
Q-2	63	( <b>n</b> n	1150/	<b>• •</b> • • •				
0-3	63	69.0	11526	3.596	41444	40588	82032	1548
Q-4 0 1	03	07 5	1/ 500		15071	(	17/00/	
Q-1 0 2	04	91.0	10580	3.913	65876	69048	134924	30866
Q-2	04	22.0	F ( ) (	2 0/5				
Q-3	04	33.0	5634	3.265	18393	21 321	39714	8210
Q-4	64 7 E	-	10/1					
<u><u>u</u>-1</u>	65	1.5	1366	3.455	4719	6509	11228	290
	62							
Q-3	00		60	4.40/	268	297	565	
тот	AL	598.5	102527		360013	369355	729368	289149

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM9PREPRODUCTION5-SUBSYSTEM90PREPRODUCTIONSUBDOFWORKTOOLINGANDSTE

		MPC	SUB Total	GδA	TOTAL COST
Q-3	59		24		24
0 - 4	<b>5</b> 9 <b>6</b> 0	53	10565	201	10766
Q-2	60			1944 - 197 - 1984 - 1974 - 1984 - 1974 - 1984 - 1974 - 1984 - 1974 - 1984 - 1974 - 1984 - 1974 - 1974 - 1974 -	a 17 a 17 a 17
Q-3	<b>6</b> 0	5572	209568	3993	213561
Q-4 Q-1	61	3041	147294	2737	150031
Q-2	61				
Q-3	<b>61</b>	6115	178507	3317	181824
Q-1	62	4414	108181	1816	109997
Q-2	62				
Q-3	62	3238	55497	932	56429
0-4	62		21027	a / <b>7</b>	22224
Q = 1 Q = 2	63 63		21937	307	22304
Q-3	63	152	83732	1400	85132
Q-4	63				
Q-1 Q-2	64 64	3290	169080	3598	172678
Q-3	64	2987	50911	1083	51994
Q-4	64				
0 - 1	65 65	88	11606	310	11916
Q-3	65		565	15	58C
тот	AL	28950	1047467	19769	1067236

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING 4-SYSTEM 9 5-SUBSYSTEM 90 PREPRODUCTION

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	60	7.5	1311	5.216	6838	4927	11765
Q-2	60						x
Q-3	60	15.0	2632	4.106	10808	9829	20637
Q-4	60						
Q-1	61	15.0	2549	3.703	9439	9633	19072
Q-2	61						•
Q-3	61	3.0	634	3.842	2436	3766	6202
Q-4	61						
Q-1	62	1.5	186	3.769	701	85 <b>9</b>	1560
Q-2	62						
Q-3	62					47	47
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	63		102	6.892	703	570	1273
Q-4	63						
Q-1	64	7.5	1340	4.669	6256	8693	14349
Q-2	64						
Q-3	64	1.5	168	5.637	947	1457	2404
тот	AL	51.0	892 <b>2</b>		38128	39181	77309

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PRODUCTION 4-SYSTEM 9 5-SUBSYSTEM 90 PREPRODUCTION

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-1 Q-2	60 60	15.0	2535	3.409	8641	11495	20136
Q-3 Q-4	60 60	147.0	24680	3.075	75894	100779	176673
Q-1 Q-2	61 61	129.0	21979	3.123	68643	89110	157753
Q-3 Q-4	61 61	46.5	8500	2.963	25188	47942	73130
Q-1 Q-2	62 62	18.0	3062	2.910	8911	17899	26810
Q-3 Q-4	62 62					594	594
Q-1 Q-2	63 63						
Q-3 Q-4	63 63	9.0	1587	4.100	6506	6180	12686
Q-1 Q-2	64 64	15.0	2502	3.133	7839	13077	20916
Q-3	64		-73	2.877	-210	-415	-625
тот	AL	379.5	64772		201412	286661	488073

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 9 5-SUBSYSTEM 90 PREPRODUCTION

		MAN- MONTHS	LABOR HOURS	LABOR Rate	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	60	12.0	2124	2.803	5953	8679	14632
Q-2	60						
Q-3	60	22.5	3793	3.079	11680	12042	23722
Q-4	60						
Q-1	61	9.0	1435	3.203	4596	4777	9373
Q-2	61						
Q-3	61		34	2.706	92	404	496
Q-4	61						
0-1	62						
Q-2	62						
Q-3	62						
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	63						
TOT	AL	43.5	7386		22321	25902	48223

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TODLING AND STE 4-SYSTEM 9 5-SUBSYSTEM 90 PREPRODUCTION

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABUR DOLLAR S	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-3	59		5	2.200	11	13	24
Q-4	59						
Q-1	60	9.0	1610	3.376	5436	4612	10048
Q-2	<b>6</b> 0						
Q-3	60	145.5	24383	3.309	80687	79804	160491
Q-4	60						
Q-1	61	94.5	16098	3.262	52518	53803	106321
Q-2	61						
Q-3	61	70.5	12904	3.860	49810	48540	98350
Q-4	61				,		
Q-1	62	39.0	6652	3.411	22691	25061	47752
Q-2	62						
Q-3	62	9.0	1464	2.794	4091	7060	11151
Q-4	62						
Q-1	63	18.0	3083	3.299	10171	11746	21917
Q-2	63						
Q-3	63	66.0	11007	3.607	39698	40.325	80023
Q-4	63						
Q-1	64	96.0	16320	3.968	64757	67 892	132649
Q-2	64						
Q-3	64	31.5	5422	3.186	17272	20190	37462
Q-4	64						
Q-1	65	7.5	1260	3.439	4333	6 <b>5</b> 09	10842
Q-2	65						
Q-3	65		60	4.467	268	297	565
тот	AL	586.5	100268		351743	365852	717595

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PLANNING

4-SYSTEM 9 5-SUBSYSTEM 90 PREPRODUCTION

	MAN- MONTHS	LABOR HOUR S	LABCR RATE	LABOR DOLLARS	BURDEN Doll Ars	LABOR + BURDEN \$
Q-3 60		8	2.625	21		21
Q-4 60						
Q-1 61		98	2.888	283		283
Q-2 61						
Q-3 61		40	2.875	115	18	133
Q-4 61						
Q-1 62		9	3.556	32	6	38
Q-2 62						
Q-3 62					1	1
Q-4 62						
Q-1 63						
0-2 63						
Q-3 63	4.5	745	3.228	2405	97	2502
Q-4 63						
Q-1 64		56	3.554	199	239	438
Q-2 64						
Q-3 64		4	5.000	20	20	40
TOTAL	4.5	960		3075	381	3456

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC 4-SYSTEM 9 5-SUBSYSTEM 90 PREPRODUCTION

(

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-1	60	4.0	646	3.125	2019		2019
Q-2	60						
Q-3	60	16.5	2871	3.422	9824		9824
Q-4	60						
Q-1	61	24.0	4036	3.184	12852		12852
Q-2	61						
Q-3	61	15.0	2635	3.388	8927		8927
Q-4	61						
Q-1	62	9.0	1626	4.188	6809		6809
Q-2	62						
Q-3	62		4	3.250	13		13
Q-4	62						
Q-1	63		5	4.000	20		20
Q-2	63						
Q-3	63	1.5	240	3.783	908		908
Q-4	63						
Q-1	64	1.5	343	5.00C	1715		1715
Q-2	64						
Q-3	64		50	3.680	184		184
Q-4	64						
Q-1	65		106	3.642	386		386
тот	AL	71.5	12562		43657		43657

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### 4-SYSTEM 9 5-SUBSYSTEM 90 PREPRODUCTION

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL AR S	LABOR + BURDEN \$	ENGR MATL
Q-3	59		5	2.200	11	13	24	
Q-4	59							
0-1	60	47.5	8226	3.512	28887	29713	58600	2596
Q-2	60							
Q-3	60	346.5	58367	3.237	188914	202454	391368	7497
Q-4	60							
Q-1	61	271.5	46195	3.211	148331	157323	305654	1295
Q-2	61				<b>-</b>			
Q-3	61	135.0	24747	3.498	86568	100670	187238	- 8
Q-4	61							
Q-1	62	67.5	11535	3.393	39144	43825	82969	17
Q-2	62							
Q-3	62	9.0	1468	2.196	4104	7702	11806	
Q-4	62	10.0	2000	2 2 2 2	10101		01 00 <b>7</b>	
Q-1	63	18.0	3088	3.300	10191	11746	21937	
0-2	6.5		12/01	2 (7)	50000	17170	07202	
Q-3	03	81.0	13081	3.071	50220	4/1/2	91392	
0-4	65	120.0	205(1	2 0 2 9	20744	90701	1700/7	
	04 67	120.0	20501	2.920	00100	09.501	110067	
Q-2	64 47	22.0	5571	2 240	10212	21 25 2	20/45	
Q-0	04 44	55.0	2571	3.209	10215	21202	57405	
0_1	46	7 5	1266	2 45 5	4710	6500	11229	
0-2	65	1.5	1500	20422	4117	0.009	11220	
w-2	65		60	4 467	26.9	207	565	
Q~-⊃	00		50		200	2.71	200	
тот	TAL	1136.5	194870		660336	717977	1378313	11413

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 9 5-SUBSYSTEM 90 PREPRODUCTION

		MFG MATL	TOOL/STE MATL	TOTAL MATERIAL	MPC	OT HER C OST	SUB TOTAL	GδA
Q-3	59						24	
Q = 4 Q = 1 Q = 2	59 60 60	585	405	3586	471		62657	1193
Q-3 Q-4	60 60	23519	42374	73390	9651	453	474862	9048
Q-1 Q-2	61 61	22200	35983	59478	5026	29	370187	6879
Q-3 Q-4	61 61	5808	72363	78179	6607	5	272029	5055
Q-1 Q-2	62 62	1933	56015	57965	4567		145501	2443
Q-3 Q-4	62 62	1516	41095	42611	3357	•	57774	970
Q-1 Q-2	63 63				,	-485	21452	359
Q-3 Q-4	63 63	6871	1548	8419	829	452	107092	1791
0-1 0-2	64 64	2778	30866	33644	3586		207297	4411
Q-3 Q-4	64 64	22	8210	8232	2987		50684	1083
Q-1 Q-2	65 65		290	290	88		11606	310
Q-3	65						565	15
тот	AL	65232	289149	365794	37169	454	1781730	33557

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# 4-SYSTEM 9 5-SUBSYSTEM 90 PREPRODUCTION

		TOTAL
Q-3	59	24
Q-4	59	
Q-1	60	63850
Q-2	60	
Q-3	60	483910
Q-4	<b>6</b> 0	
Q-1	61	377066
Q-2	61	
Q-3	61	277084
Q-4	61	
Q-1	62	147944
Q-2	62	
Q-3	62	58744
Q-4	62	
Q-1	63	21811
Q-2	63	
Q-3	63	108883
Q-4	63	
Q-1	64	211708
Q-2	64	
Q-3	64	51767
Q-4	64	
Q-1	65	11916
Q-2	65	<b></b>
Q-3	65	580
тот	TAL	1815287

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## COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 9 5-SUBSYSTEM 91 TRAINING

	DESIGN /ENGR HOURS DOLLARS	PROD TOTAL Hours Hours Dollars Dollars
DESIGN/ENGINEERING	104274	104274
LABOR AT \$ 4.436	462569	462569
ENGR BURDEN AT \$ 3.849	401368	401368
ENGR MATERIAL	1434	1434
MPC	136	136
OTHER COST .	903	903
	محمد مرديه وسرب ويوب مشمة بهويد بيجيه فالتله	
SUB-TOTAL	866410	866410
GEN & ADMIN	2310	2310
TOTAL COST	868720	868720

TIME-PHASED COST DETAIL-SEE PAGE II-708

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING

4-SYSTEM	9	
5-SUB SYSTEM	91	TRAINING
SUBD OF WORK	DESIGN/ENGINEERI	NG

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-1 Q-2	58 58	27.0	4510	4.273	19271	20519	39790
Q-3 Q-4	58 58	234.0	39429	4.250	167576	156883	324459
Q-1 Q-2	59 59	184.5	31522	4.285	135062	115229	250291
Q-3 Q-4	59 59	87.0	15291	4.586	71647	57227	128874
0-1 0-2	60 60	42.0	7279	4.905	35700	26827	62527
Q-3 Q-4	60 60		46	6.217	286	55	341
$\overline{Q-1}$ $\overline{Q-2}$	61 61	34.5	5850	5.433	31784	19836	51620
Q-3 Q-4	61 61		75	4.107	308	3275	3583
Q-1 Q-2 Q-3	62 62 62						
Q-4 Q-1	62 63	1.5	272	3.438	935	1477	2412
Q-2 Q-3	63 63					40	40
тот	AL	610.5	104274		462569	401 368	863937

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 9 5-SUBSYSTEM 91 SUBD OF WORK DESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN Doll Ars	LABOR + BURDEN \$	ENG MAT
Q-1 Q-2	58 58	27.0	4510	4.273	19271	20519	39790	
Q-3 Q-4	58 58	234.0	39429	4.250	167576	156883	324459	
Q-1 Q-2	59 59	184.5	31522	4.285	135062	115229	250291	
Q-3 Q-4	59 59	87.0	15291	4.686	71647	57 227	128874	
Q = 1 Q = 2	60 60	42.0	7279	4.905	35700	26 827	62527	
Q-3 Q-4	60 60		46	6.217	286	55	341	
Q-1 Q-2	61 61	34.5	5850	5.433	31784	19836	51620	
Q-3 Q-4 Q-1	61 61 62		75	4.107	308	3275	3 <b>583</b>	
Q-2 Q-3	62 62							27
Q-4 Q-1	62 63	1.5	272	3.438	935	1477	2412	154
0-3	63					40	40	-38
TOT	AL	610.5	104274		462569	401368	863937	143

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TRAINING

4-SYSTEM 9 5-SUBSYSTEM 91 SUBD OF WORK DESIGN/ENGINEERING

.

	MPC	OTHER COST	SUB Total	G&A	TOTAL COST
Q-1 50	3		39790		39790
	5				
Q-3 50	5		324459		324459
Q-4 50	5				
			250291		250291
Q-2 5%	1				
Q-3 5:	? }		128874		128874
Q - 4 5	j .				
		815	63342	1207	64 54 9
Q = 2  60	}				
	)		341	6	347
Q-4 6(		0.0			_
		88	51708	961	52669
0 - 2 0					
		506	4089	76	4165
Q-4 01		50/	<b>F</b> = 1	_	
		-506	-506	-8	-514
Q-2 02				-	
		11	308	5	313
Q-4 62	150				
Q−1 0: 0 2 / 2	152		4112	69	4181
	20				
<b>u-3</b> 03	-38	-11	-398	-6	-404
TOTAL	136	903	866410	2310	868720

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 9 5-SUBSYSTEM 91 TRAINING

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + Burden \$	ENGR MATL
Q-1 Q-2	58 58	27.0	4510	4.273	19271	20519	39790	
Q-3 Q-4	58 58	234.0	39429	4.250	167576	156883	324459	
$\hat{Q} = 1$ $\hat{Q} = 2$	59 59	184.5	31522	4.285	135062	115229	250291	
Q-3 0-4	59 59	87.0	15291	4.686	71647	57227	123874	
Q-1 y-2	60 60	42.0	7279	4.905	35700	26827	62527	
Q-3 Q-4	60 60		46	6.217	286	55	341	
0 - 1 0 - 2	61 61	34.5	5850	5.433	31784	19836	51620	
Q = 3 Q = 4 Q = 1 Q = 2	61 61 62		75	4.107	308	3275	3583	
Q-2 Q-3	62							275
Q-4 Q-1 Q-2 Q-2	62 63 63	1.5	272	3.438	935	1477	2412	1548
с-9 гот	сэ • Л1	610 5	104274		462540	40	942027	-389
101	~L	010.0	107214		402007	401 200	002751	1434

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## 4-SYSTEM 9 5-SUBSYSTEM 91 TRAINING

.

		MPC COST	SUB TOTAL	GEA	TOTAL COST
0-1	58		39790		39790
Q-2	58				
Q-3	58		324459		324459
0-4	58				
Q-1	59		250291		250291
Q-2	59				
Q-3	59		128874		128874
Q-4	59				
Q-1	60	815	63342	1207	64549
Q-2	50				
Q-3	60		341	6	347
Q-4	60				
Q-1	61	88	51708	961	52669
Q - 2	61				
Q-3	61	506	4089	76	4165
Q-4	61				
Q-1	62	-506	-506	-8	-514
Q-2	62				
Q-3	62	22 11	308	5	313
Q-4	62				
Q-1	63	152	4112	69	4181
Q-2	63				
Q-3	63	-38 -11	-398	-6	-404
тот	AL	136 903	866410	2310	868720

## COST BREAKDOWNS B-70 AIRCRAFT STUDY

	DESIGN /ENGR	TOTAL
	HOURS	HOURS
	DOLLARS	DOLLARS
DESIGN/ENGINEERING	210947	210947
LABOR AT \$ 4.589	968036	968036
ENGR BURDEN AT \$ 5.240	1105284	1105284
ENGR MATERIAL	129711	129711
MPC	21244	21244
OTHER COST	<u> </u>	39254
SUB-TOTAL	2263529	2263529
GEN & ADMIN	46767	46767
TOTAL COST	2310296	2310296

TIME-PHASED COST	
DETAIL-SEE PAGE	II-714

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING 4-SYSTEM 9 5-SUBSYSTEM 92 SUBD OF WORK DESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
0-1	58	3.0	443	3.898	1727	2012	3739
Q-2	58						
Q-3	58	1.5	169	3.740	632	457	1089
Q-4	58						
Q-1	59	4.5	771	3.949	3045	2739	5784
Q-2	59						
Q-3	59	22.5	4069	3.969	16149	16 127	32276
Q-4	59						
Q <b>-1</b>	60	16.5	2816	4.452	12536	12197	24733
Q-2	60						
Q-3	60	18.0	2951	4.456	13151	12403	25554
0-4	60						
Q-1	61	27.0	4626	4.473	20692	18280	38972
Q = Z	61	20.0					
Q-3	01	30.0	5403	4.738	25598	24805	50403
0 1	61	E 0 <i>E</i>	0000				
	02	28.5	9899	4.719	46718	43354	90072
0-2	0Z 40	170 5	20012	/ 100	100710	100000	
0-4	02 60	110.0	30012	4.157	125719	130973	256692
0 - 1	62	201 0	24204	1 222	140000	150770	207075
0-2	63	201.0	24204	4.000	140202	123113	301915
$\tilde{0} - 3$	63	168-0	28165	4 479	126154	1/2710	2609 <b>7</b> 2
0-4	63	100.0	20105	T • T 1 2	160104	142110	209012
$\Omega - 1$	64	145-5	24784	4.556	112912	120425	742227
$\tilde{0}-2$	64	11303			116716	100460	270001
$\overline{0}-3$	64	111.0	19476	5.386	104892	120510	225402
Q-4	64		ೆ ಕಾಲ್ಗಳ	~~~~	<b>XV</b> • <i>V</i> > <b>C</b>	ILUJIU	662792
Q-1	65	58.5	10064	3.825	38491	47321	85812
Q-2	65				and a second	· · · · · · · · · · · · · · · · · · ·	المعلى المعلم

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

DESIGN/ENGINEERING 4-SYSTEM 9 5-SUBSYSTEM 92 PUBLICATIONS SUBD OF WORK DESIGN/ENGINEERING

#### ON-SITE LABOR

	MAN- MON THS	LABOR HOUR S	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 65 Q-4 65	36.0	6013	5.445	32743	39002	71745
Q-1 66	66.0	11494	5.196	59719	86 11 0	145829
Q-3 66	93.0	15588	5.065	78956	115078	194034
TOTAL	1239.0	210947		968036	1105284	2073320

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### 4-SYSTEM 9 5-SUBSYSTEM 92 PUBLICATIONS

		MAN- MON TH S	LABOR HOUR S	LABOR RATE	LABUR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR Matl
Q-1	58	3.0	443	3.898	1727	2012	3739	
Q-2	58 50	1 6	1.00	2 7/ 6	( ) .			
Q-3	28 58	1+5	163	3.140	632	457	1089	
Q-1	59	4.5	771	3.949	3045	2739	5784	
Q-2	59							
Q-3	59	22.5	4069	3.969	16149	16127	32276	
Q - 4	59	16 5	2014	4 45 3	10504	10107		· • • • •
Q = 2	60	10.5	2010	4.422	12000	12197	24733	321
Q-3	60	18.0	2951	4.456	13151	12403	25554	213
Q-4	60							
	61	27.0	4626	4.473	20692	18280	38 <b>972</b>	734
Q-2 0-3	61	30.0	5403	4 73.8	25509	24 80 E	50403	11.04
0-4	61		2405	4.10	23390	24003	20403	4400
Q-1	62	58.5	9899	4.719	46718	43354	90072	3940
Q-2	62							
0-3	62	178.5	30012	4.189	125719	130973	256692	17939
0-4 0-1	02 63	201.0	34204	4.333	148202	150773	307075	6657
Q-2	63	2010 ···	54204	<b>T U</b> U U U	LTULUL	122112	201012	0077
2-3	63	168.0	28165	4.479	126154	143718	269872	24085
Q-4	63						- ·	
	64 64	145.5	24184	4.556	112912	130425	243337	5549
2-2	64	111.0	19476	5.386	104892	120510	225402	13898
2-4	64							
2-1	65	58.5	10064	3.825	38491	47321	85812	14626
Q-2	65 (E	24.0	(0) 0	5 115		20.000		1/1/0
2-3	02 65	30.0	0013	2.445	32143	39002	11745	14142

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## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	MAN- MON THS	LABOR HOURS	LABOR RATE	LABUR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
Q = 1 66 Q = 2 66	66.0	11494	5.196	59719	86110	145829	13797
Q-3 66	93.0	15588	5.065	78956	115078	194034	9318
TOTAL	1239.0	210947		968036	1105284	2073320	129711

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

		MPC	OTHER COST	SUB Total	G&A	TOTAL Cost
Q-1	58			3739		3 <b>7</b> 39
Q-2	58					
Q-3	58			1089		1089
Q-4	58					
Q-1	59			5784		5784
Q-2	59					
Q-3	59			32276		32276
Q-4	59					
Q-1	60	43	172	25275	482	25757
Q-2	60					
Q-3	60	28		25795	491	26286
Q-4	60					
Q-1	61	62	495	40263	748	41011
Q-2	61					
Q-3	61	379		55268	1027	56295
Q-4	61					
Q-1	62	310	1638	95 96 0	1611	97571
Q-2	62					
Q-3	62	1414	3834	279879	4698	284577
Q-4	62				•	
Q-1	63	656	4334	319622	5344	324966
0-2	63					
Q-3	63	2372	4039	300368	5022	305390
Q-4	63					
Q-1	64	592	4064	253542	5395	258937
Q-2	64					
Q-3	64	5056	5247	249603	5311	254914
Q-4	64					
Q-1	65	4375	2208	107021	2855	109876
Q-2	65					
Q-3	65	2523	2602	91012	2428	93440
Q-4	65					

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	MPC	OTHER COST	SUB TOTAL	G&A	TOTAL COST
Q = 1 66 Q = 2 66	2861	5472	167959	5058	173017
Q-3 66	573	5149	209074	6297	215371
TOTAL	21244	39254	2263529	46767	2310296

#### COST BREAKDOWNS B-70 AIRCRAFT STUDY

#### 4-SYSTEM 9 5-SUBSYSTEM 93 PHOTOGRAPHIC LAB

	DESIGN /ENGR HOURS DOLLARS	TOTAL Hours Dollars
DESIGN/ENGINEERING	46644	46644
LABOR AT \$ 4.944	230590	230590
ENGR BURDEN AT \$ 4.384	204502	204502
	and the states and a spin states again.	
SUB-TOTAL	435092	435092
GEN & ADMIN	7168	7168
TOTAL COST	4422.60	442260

TIME-PHASED COST DETAIL-SEE PAGE II-721

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

DESIGN/ENGINEERING 4-SYSTEM 9 PHOTOGRAPHIC LAB 5-SUBSYSTEM 93 SUBD OF WORK DESIGN/ENGINEERING

		MAN- MON TH S	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	55	6.0	923	4.514	4166	3174	7340
Q-2	59						
Q-3	59	21.0	3663	4.144	15179	13144	28323
Q-4	59						
Q-1	60	33.0	5771	4.508	26017	21243	47260
Q-2	60						
Q-3	60	37.5	6224	4.756	29604	22765	52369
Q-4	60						
Q-1	61	46.5	7960	4.777	38024	35064	73088
Q-2	61						
Q-3	61	27.0	4994	5.050	25219	23897	49116
Q-4	61			· .			
Q-1	62	45.0	7696	5.358	41234	35423	76657
Q-2	62						
Q-3	62	37.5	6338	5.290	33526	32484	66010
Q-4	62						
Q-1	63	6.0	1040	5.663	5889	5569	11458
Q-2	63						
Q-3	<b>6</b> 3	10.5	1678	5.740	9631	9484	19115
Q-4	63				•		
Q-1	64	1.5	354	5.884	2083	2236	4319
Q-2	64						
Q-3	64		3	6.000	18	19	37
тот	AL	271.5	46544		230590	204502	435092

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### 4-SYSTEM 9 5-SUBSYSTEM 93 PHOTOGRAPHIC LAB

		MAN- MONTHS	LABOR	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$	GEA
Q-1	59 50	6.0	923	4.514	4166	3174	7340	
Q = Z	59	<b>a a a</b>		, , , ,		10111		
Q-3	22	21.0	2002	4.144	15179	13144	28323	
$\sqrt{-4}$	59 40	22 0	5771	1 500	26017	31 34 3	17040	000
Q−1 ∩_2	60 60	3 <b>3</b> •0	2771	4.000	20017	21245	47200	900
Q = 2 Q = 3	60 60	37.5	6224	4.756	29604	22765	52369	998
Q-4	60							
Q-1	61	46.5	7960	4.777	38024	35064	73088	1358
Q-2	61							
Q-3	61	27.0	4994	5.050	25219	2389 <b>7</b>	49116	913
Q-4	61							
Q-1	62	45.0	7696	5.358	41234	35423	76657	1287
Q-2	62	_						
Q-3	62	37.5	6338	5.290	33526	32484	66010	1108
Q-4	62							
Q-1	63	6.9	1040	5.663	5889	5569	11458	192
Q-2	63	10 5	1/70		0/01	<b>0</b> /0/		~~~
Q-3	63	10.5	1678	5.140	9531	9484	19112	320
Q-4	63	<b>3</b> m	251	5 004	2000	222/	(210	0.2
Q-1	04	1.0	304	2.004	2065	2230	4019	72
0-2	04 4 /		2	4 000	10	10	27	
Q-3	04		э.	0.000	10	19	21	
тот	AL	271.5	46644		230590	204502	435092	7168

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 9 5-SUBSYSTEM 93 PHOTOGRAPHIC LAB

		TOTAL Cost
Q <b>-</b> 1	<b>.</b> 59	7340
Q-2	59	
Q-3	59	28323
Q-4	59	
Q-1	60	48160
Q-2	60	
Q-3	60	53367
Q-4	60	
Q-1	61	74446
Q-2	61	
Q-3	61	50029
Q-4	61	
Q-1	62	77944
Q-2	62	
Q-3	62	67118
Q-4	62	
Q-1	63	11650
Q-2	63	
Q-3	63	19435
Q-4	63	
Q-1	64	4411
Q-2	64	
Q-3	64	37
тот	<b>FAL</b>	442260

## COST BREAKDOWNS B-70 AIRCRAFT STUDY

## 4-SYSTEM 9 5-SUBSYSTEM 94 RELIABILITY

	DESIGN /ENGR HOURS DOLLARS	TOTAL HOURS DOLLARS
DESIGN/ENGINEERING LABCR AT \$ 4.994 ENGR BURDEN AT \$ 4.702	83785 418454 393933	83785 418454 393933
SUB-TOTAL	812387	812387
GEN & ADMIN	13499	13499
TOTAL COST	825386	825886

TIME-PHASED COST DETAIL-SEE PAGE II-725

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	DESIGN/ENGINEER	ING
4-SYSTEM	9	TOTOT WANDER WINE
5-SUB SYSTEM	94	RELIADILITI
SUBD OF WORK	DESIGN/ENGINEER	ING

## ON-SITE LABOR

		MAN-	LABOR	LABOR	LABOR	<b>BURDEN</b>	LABOR +
		MONTHS	HOURS	RATE	DOLLARS	DOLLARS	BURDEN \$
Q-3	58	3.0	474	4.430	2100	1865	3965
Q-4	58						
Q-1	59	22.5	385 <b>7</b>	5.955	22967	13256	36223
Q-2	59						
Q-3	59	34.5	5955	4.193	24971	21462	46433
Q-4	59						
Q-1	<b>6</b> 0	39.0	6641	4.565	30319	24822	55141
Q-2	60						
Q-3	60	49.5	8216	4.773	39217	30245	69462
Q-4	60						
Q-1	61	52.5	8920	4.723	42126	29956	72082
Q-2	61						
Q-3	61	27.0	4863	5.016	24394	22241	46635
Q-4	61		~			From states (1999) - Barry	10000
Q-1	62	34.5	5977	5.218	31188	27443	58631
Q-2	62						
Q-3	62	44.5	7432	5.104	37933	37136	75069
Q-4	62						
Q-1	63	51.0	8707	5.178	45083	47010	92093
Q-2	63						- • · · -
Q-3	63	57.0	9499	5.188	49278	54258	103536
Q-4	63						
Q-1	64	40.5	6843	4.967	33987	41265	75252
Q-2	64						
Q-3	64	24.0	4160	5.395	22445	28 2 3 1	50676
Q-4	64						
Q-1	65	7.5	1305	5.534	7222	8575	15797
Q-2	65						
Q-3	65	6.0	936	5.581	522.4	6168	11392
тот	AL	493.0	83785		418454	393933	812387

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### 4-SYSTEM 9 5-SUBSYSTEM 94 RELIABILITY

	MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	GEA	×
Q-3 58	3.0	474	4.430	2100	1865	3965		
Q-4 58	3							
Q-1 59	22.5	385 <b>7</b>	5.955	2296 <b>7</b>	13256	36223		
Q-2 59	)							
Q-3 59	34.5	5955	4.193	24971	21.462	46433		
Q-4 55		1213		0.001.0	24 0 2 2	87 6 <b>3</b> 7 <b>3</b>	1061	
	) <u>3</u> 9•0	6641	4.505	30319	24822	52141	1001	
Q = 2 60 Q = 3 60	, ) 49.5	8216	4.773	39217	30245	69462	1323	
Q-4 60	)							E
Q <b>-1</b> 61	. 52.5	8920	4.723	42126	29956	72082	1339	
0-2 61								
Q-3 61	. 27.0	4863	5.016	24394	22 241	46635	867	
Q-4 61	•							
Q - 1 62	34.5	597 <b>7</b>	5.218	31188	27443	58631	984	
Q-2 62					17 mm a un 2	····•		
Q-3 62	44.5	7432	5.104	37933	37136	15069	1260	
Q - 4 62		0707	r 170	15000	(7010	02002	1540	
Q-1 01	5 51.0	8707	5.118	42083	47010	92093	1040	
V-Z 03 V-2 43	) I 57 ()		5 199	49278	54 25 8	103536	1.731	
Q-3 0: 0-4 67	) () () () () () () () () () () () () ()	7477	2.100	47210	54250	102220	T.4 "T	
Q = 4 0	, + 40.5	6843	4.967	33987	41265	75252	1601	
0-2 64	ł							
0-3 64	e 24.0	4160	5.395	22445	28231	50676	1078	
Q-4 64	•							
Q-1 65	5 <b>7.</b> 5	1305	5.534	7222	8575	15797	421	
Q-2 65	5							
Q-3 65	5 6.0	936	5.581	5224	6168	11392	304	
TOTAL	. 493.0	83785		418454	393933	812387	13499	

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM	9
5-SUB SYSTEM	94
RELIABILI	(TY

		TOTAL COST
Q-3	58	3965
Q = 4 Q = 1	58 50	24 2 2 2
$\hat{0} = 2$	59	30225
() - 3	59	46433
0-4	59	
Q-1	60	56192
Q-2	60	
ର–3	<b>6</b> 0	70785
Q-4	60	
Q <b>-1</b>	61	73421
Q-2	61	
0-4	01	47502
0-1	01 67	50415
0-2	62	59015
0-3	62	76329
0-4	62	a tar un han P
Q-1	63	93633
Q-2	63	
Q-3	63	105267
0-4	63	
0-1	64	76853
0-2	64	51754
0-4	04 64	21724
9-4 0-1	65	16218
$\sqrt[3]{-2}$	65	19210
0-3	65	11696
TOT	AL	825886

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## COST BREAKDOWNS B-70 AIRCRAFT STUDY

#### 4-SYSTEM 9 5-SUBSYSTEM 96 COMPUTING

·	DESIGN /ENGR HOURS DOLLARS	TOTAL Hours Dollars
DESIGN/ENGINEERING	194434	194434
LABOR AT \$ 5.171	1005464	1005464
ENGR BURDEN AT \$ 4.687	911337	911337
SUB-TOTAL	1916801	1916801
GEN & ADMIN	31522	31522
TOTAL COST	1948323	1948323

TIME-PHASED COST DETAIL-SEE PAGE II-729

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING

4-SYSTEM 9 5-SUBSYSTEM 96 SUBD OF WORK DESIGN/ENGINEERING

## ON-SITE LABOR

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		MAN-	LABOR	LABOR	LABOR	<b>BUR DEN</b>	LABOR +
		MONTHS	HOURS	RATE	DOLLARS	DOLLARS	BURDEN \$
Q-1	58	13.5	2168	4.558	9882	9864	19746
Q-2	58						
Q-3	58	42.0	7078	4.351	30795	27671	58466
Q-4	58						
Q-1	59	51.0	8623	4.336	37386	29568	66954
Q-2	59						
Q-3	59	57.0	9986	4.245	42386	36105	78491
Q-4	59						
Q-1	60	58.5	10263	4.563	46830	38265	85095
Q-2	60						
Q-3	60	99.0	16522	4,733	73194	60775	138969
Q-4	60						
Q-1	61	141.0	24091	4.744	114278	81158	195436
Q-2	61						```
Q-3	61	81.0	14819	4.999	74082	67923	142005
Q-4	61						
Q-1	62	127.5	21701	5.279	114566	99810	214376
Q-2	62						·
Q-3	62	159.0	26643	5.351	142555	107096	249651
Q-4	62						
Q-1	63	112.5	19231	6.228	119778	123590	243368
Q-2	63						
Q-3	63	60.0	10133	5.390	54616	61776	116392
0-4	63						
Q-1	64	51.0	8602	5.747	49437	54842	104279
0-2	64	•					
Q-3	64	40.5	7167	5.837	41835	46220	88055
Q-4	64						
Q-1	65	36.0	6342	6.579	41727	45839	87566
Q-2	65						

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

# DESIGN/ENGINEERING 4-SYSTEM 9 COMPUTING 5-SUBSYSTEM 96 COMPUTING SUBD OF WORK DESIGN/ENGINEERING

	MAN- MON TH S	LABOR HOUR S	LABOR RATE	LABOR DGLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3 65 Q-4 65	6.0	1065	6.683	7117	19751	26868
Q-1 66					1084	1084
TOTAL	1135.5	194434		1005464	911337	1916801

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### 4-SYSTEM 9 5-SUBSYSTEM 96 COMPUTING

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
0 - 1	58 58	13.5	2168	4.558	9882	9864	19746	
Q-3 0-4	58 58	42.0	7078	4.351	30795	27671	58466	
Q-1 Q-2	59 59	51.0	8623	4.336	37386	29568	66954	
Q-3 Q-4	59 59	57.0	9986	4.245	42386	36105	78491	
Q-1 Q-2	60 60	58.5	10263	4.563	<b>4683</b> 0	38265	85095	
Q-3 Q-4	60 60	99.0	16522	4.733	78194	60775	138969	
Q-1 Q-2	61 61	141.0	24091	4.744	114278	81158	195436	
Q-3 Q-4	61 61	81.0	14819	4.999	74082	67923	142005	
Q-1 Q-2	62 62	127.5	21701	5.279	114566	\ <b>9981</b> 0	214376	
Q-3 Q-4	62 62	159.0	26643	5.351	142555	107096	249651	
Q-1 Q-2	63 63	112.5	19231	6.228	119778	123590	243368	
Q-3 Q-4	63 63	60.0	10133	5.390	54616	61776	116392	
Q-1 Q-2	64 64	51.0	8602	5.747	4943 <b>7</b>	54842	104279	
Q-3 Q-4	64 64	40.5	7167	5.837	41835	46220	88055	
Q-1 Q-2	65 65	36.0	6342	6.579	41727	<b>45</b> 839	87566	
Q-3 Q-4	65 65	6.0	1065	6.683	7117	19751	26868	

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## 4-SYSTEM 9 5-SUBSYSTEM 96 COMPUTING

	MAN- MONTHS	LABOR HOURS	LABUR RATE	LABOR DOLLARS	BUR DEN DGLL ARS	LABOR + BURDEN \$	ENGR MATL
Q-1 66					1084	1084	
TOTAL	1135.5	194434		1005464	911337	1916801	

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 9 5-SUBSYSTEM 96 COMPUTING

	SUB TO TAL	GδA	TOTAL COST
<b>Q-1</b> 58	19746		19746
Q = 3 58 Q = 4 58	58466		58466
Q = 1 59 Q = 2 59	66954		66954
Q-3 59 Q-4 59	78491		78491
Q-1 60 Q-2 60	85095	1621	86716
Q-3 60 Q-4 60	138969	2648	141617
Q-1 61 Q-2 61	195436	3632	199068
0-3 61 Q-4 51	142005	2639	144644
0-1 62 Q-2 62	214376	3598	217974
Q-3 62 Q-4 62	249651	4190	253841
Q-1 63 Q-2 63	243368	4069	247437
Q-3 63 Q-4 63	116392	1946	118338
Q-1 64 Q-2 64	104279	2219	106498
0-3 64 0-4 64	88055	1874	89929
Q-1 65 Q-2 65	87566	2336	89902
Q-3 55 Q-4 65	26868	717	27585

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

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## 4-SYSTEM 9 5-SUBSYSTEM 96 COMPUTING

	SUB TO TAL	G&A	TOTAL COST
Q-1 66	1084	33	1117
TOTAL	1916801	31522	1948323

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 9 5-SUBSYSTEM 97 PHCTO-TEMPLATE LAB

	DESIGN	
	/ENGR	TOTAL
	HOURS	HOURS
	DOLLARS	DOLLARS
DESIGN/ENGINEERING	28164	28164
LABOR AT \$ 4.863	136957	136957
ENGR BURDEN AT \$ 4.090	115198	115198
	والمراجع وال	ander verste some verste vange verste deleter ander
SUB-TCTAL	252155	252155
GEN & ADMIN	3543	3543
TOTAL COST	255698	255698

TIME PHASED COST DETAIL-SEE PAGE II-736

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## DESIGN/ENGINEERING

4-SYSTEM95-SUBSYSTEM97SUBD OF WORKDESIGN/ENGINEERING

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58		94	4.670	439	428	867
Q-2	58						
Q-3	58	7.5	1234	4.625	5707	4832	10539
Q-4	58						
Q-1	59	16.5	2878	4.513	12989	9895	22884
Q-2	59						
Q-3	59	16.5	2964	4.320	12805	10633	23438
Q-4	59					-	
Q-1	60	16.5	2745	4.451	12217	10471	22688
Q-2	60						
Q-3	<b>6</b> 0	27.0	4446	4.656	20700	16557	37257
Q-4	60						57257
Q-1	61	25.5	4454	4.826	21496	15397	26893
Q-2	61				~~~~~~	و ۲۰۰۷ میں بیک	50055
Q-3	61	10.5	1889	4.852	<u>`9165</u>	8635	17800
Q-4	61						11000
Q-1	62	21.0	3459	5.328	18430	15963	24202
Q-2	62						<b>.</b>
Q-3	62	10.5	1870	5, 382	10065	9627	19692
Q-4	62				20002	JULI	LJUJZ
0-1	63	4.5	843	6.609	5571	5147	10719
0-2	63			0.000		7141	10110
0-3	63	3.0	477	5.300	2528	2475	5002
0-4	63				2720	2415	2002
0 - 1	64	3.0	536	5,979	3205	2367	6572
Q-2	64		220	3.719	2202	3301	0512
Q-3	64	1.5	275	5.964	1640	1771	3411
тот	AL	163.5	28164		136957	115198	252155

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

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4-SYSTEM 9 5-SUBSYSTEM 97 PHOTO-TEMPLATE LAB

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	G&A
0-1	58		94	4.67.0	439	428	867	
Q-2	58							
Q-3	58	7.5	1234	4.625	5 <b>7</b> 07	4832	10539	
Q-4	58							
Q-1	59	16.5	2878	4.513	12989	9895	22884	
Q-2	59							
Q-3	59	16.5	2964	4.320	12805	10633	23438	
Q-4	59							
Q-1	60	16.5	2745	4.451	12217	10471	22 <b>6</b> 88	432
Q-2	60							
Q-3	60	27.0	4446	4.656	20700	16557	37257	710
Q-4	60	25 5		1 00 1		at ans ar, an and		
Q-1 0 0	01	23.5	4454	4.825	21496	15397	36893	686
Q-2	61	10 5	1000	1 05 0	01/5	0/25	17000	
	C1 61	10.5	1893	4.072	9100	8035	17800	166
0-1	62	21.0	3450	5 270	19420	15042	2/202	577
0 - 2	62	2100	2427	1.20	10450	10905	24070	211
0-3	62	10.5	1870	5, 392	10065	9627	19692	331
Q-4	62	10.0	1010	J. 302	14042	2021	17072	224
0-1	63	4.5	843	6.509	5571	5147	10718	179
0-2	63		5.5		san i san marina anna anna anna anna anna anna anna	und" dan 1 1		
Q-3	63	3.0	477	5.300	2528	2475	5003	84
Q-4	63							
Q-1	64	3.0	536	5.979	3205	3367	6572	140
Q-2	64							
Q-3	64	1.5	275	5.964	1640	1771	3411	73
тот	AL	163.5	28164		136957	115198	252155	3543

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

## 4-SYSTEM 9 5-SUBSYSTEM 97 PHOTO-TEMPLATE LAB

		TOTAL
		COST
0-1	53	867
$\tilde{0}-2$	58	001
0-3	58	10530
0-4	58	
0-1	59	22884
0-2	59	and the set of the
0-3	59	23438
Q-4	59	20100
Q-1	60	<b>231</b> 20
0-2	60	
Q-3	60	37967
Q-4	60	
Q-1	61	37579
Q-2	61	
Q-3	61	18131
Q-4	61	
Q-1	62	34970
Q-2	62	
Q-3	62	20023
Q-4	62	
0-1	63	10897
0-2	63	
0-3	63	5087
Q-4	63	
Q-1	64	6712
Q-2	64	
Q-3	64	3484
тот	AL	255698

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 9 5-SUBSYSTEM 98

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TERMINATION COSTS

	DESIGN		TOOLING	OTHER
	/ENGR	PROD	AND STE	COST
	HOURS	HOURS	HOUR S	HOURS
	DOLLARS	DOLLARS	DOLLARS	DOLLARS
OTHER COST	1989913	291299	14383	5118952
SUB-TOTAL	1989913	291299	14383	5118952
GEN & ADMIN	43715	6074	322	109695
TOTAL COST	2033628	297373	14705	5228647
ТТ	ME-PHASED COSE			

TIME-PHASED COST			
DETAIL-SEE PAGE	II-741	II-742	II-743

- 7

## COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM	9	······································	
5-SUB SYSTEM	98	TERMINATION (	COSTS

	TOTAL HOURS
	DOLLARS
OTHER COST	7414547
SUB-TOTAL	7414547
GEN & ADMIN	159806
TOTAL COST	7574353

TIME-PHASED COST DETAIL-SEE PAGE

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM9TERMINATION COSTS5-SUBSYSTEM98SUBD OF WORKDESIGN/ENGINEERING

.

		ENGR OTHER MATL COST	SUB TOTAL	G&A	TOTAL COST
Q-1	<b>6</b> 0	592	592	11	603
Q-2 Q-3	60 60	15767	15767	300	16067
Q-1	6U 61	35483	35483	659	36142
Q=2 Q=3	61 61	500290	500290	9297	509587
$Q^{-1}$ $Q^{-1}$	62 62	367130	367130	6162	373292
Q-3 0-4	62 62	333497	333497	5598	339095
Q-1 Q-2	63 63	9988 <b>2</b>	99882	1670	101552
Q-3 Q-4	63 63	-309305	-309305	-5171	-314476
Q-1 Q-2	64 64	3690	3690	` 79	3769
Q-3 Q-4	64 64	13275	13275	282	13557
Q-1 Q-2	65 65	4324	4324	115	4439
Q-3 Q-4	65 65	917453	917453	24477	<b>9419</b> 30
Q-1 Q-2	66 66	7103	7103	214	7317
Q-3	66	732	732	22	754
TOT	<b>AL</b>	1989913	1989913	43715	2033628

#### TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

TERMINATION COSTS

4-SYSTEM 9 5-SUBSYSTEM 98 SUBD OF WORK PRODUCTION

		SUBC	OTHER COST	SUB Total	<b>G &amp; A</b>	TOTAL COST
Q-1	60		12198	12198	232	12430
Q-3	60		3642	3642	69	3711
Q-1 Q-2	61 61		1862	1862	35	1897
Q-2 Q-3	61 61		10008	10008	186	10194
Q=1 Q=2	62 62		2478	2478	42	2520
Q-2 Q-3	62 62		1809	1809	30	1839
Q-1 Q-2	63 63		1179	1179	2.0	1199
Q-3 0-4	63 63		70755	70755	1183	71938
0 - 1 0 - 2	64 64		78103	78103	`1662	79765
Q-3 Q-4	64 64		65291	65291	1389	66680
Q = 1 Q = 2	65 65		21735	21735	580	22315
Q-3 Q-4	65 65	,	7157	7157	191	7348
Q-1 Q-2	<b>6</b> 6		7261	7261	219	7480
Q-3	66		7821	7821	236	8057
тот	AL		291299	291299	6074	297373

TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM9TERMINATION COSTS5-SUBSYSTEM98SUBD OF WORK TOOLING AND STE

		OTHER		TOTAL	
		COST	GξA	COST	
Q-1	60	2224	42	2266	
Q-2	60				
Q-3	60	46	1	47	
Q-4	60				
Q-1	61				
Q-2	61				
Q-3	61	1162	22	1184	
0-4	61				
Q-1	62				
Q-2	62				
Q - 3	62				
Q-4	62				
0 - 1	63				
Q-2	63				
Q-3	63				
Q-4	63				
Q-1	64	3279	70	3349	
Q-2	64				
Q-3	64	4987	106	5093	
Q-4	64				
Q-1	65				
Q-2	65				
Q-3	65				
0-4	65				
Q-1	66	1070	32	1102	
Q-2	66				
Q-3	66	1615	49	1654	
TOT	TAL	14383	322	14705	

#### TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

TERMINATION COSTS

4-SYSTEM 9 5-SUBSYSTEM 98 SUBD OF WORK OTHER COST

		SUBC	OTHER COST	SUB TOTAL	6 <b>8</b> A	TOTAL COST
Q - 1	60 60		18116	18116	345	18461
Q = 2 Q = 3	60 60		112353	112353	2141	114494
Q = 1 Q = 2	61 61		66498	66498	1236	67734
Q-3 0-4	61 61		318024	318024	5910	323934
Ω−1 Ω−2	62 62		35864	35864	60 <b>2</b>	36466
Q-3 0-4	62 62		11976	11976	201	12177
Q-1 Q-2	63 63		569 <b>7</b>	569 <b>7</b>	95	5792
Q-3 Q-4	63 63		1598681	1598681	26730	1625411
Q-1 Q-2	64 64		256870	256870	`5466	262336
Q-3 Q-4	64 64		869076	869076	1.8492	887568
Q-1 Q-2	65 65		163492	163492	4362	167854
Q-3 Q-4	65 65		1730851	1730851	46179	1777030
Q-1 Q-2	65 66		-168921	-168921	-5087	-174008
Q-3	66		100375	100375	3023	103398
TOT	TAL		5118952	5118952	109695	5228647

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 9 5-SUBSYSTEM 98

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TERMINATION COSTS

	ENGR MATL	SUBC	TOTAL MATERIAL	OTHER COST	SUB TOTAL	G&A	TOTAL COST
Q-1 60				33130	33130	630	33760
Q-2 60							
Q-3 60				131808	131808	2511	134319
Q-4 60							
Q-1 61				103843	103843	1930	105773
Q-2 61							
Q-3 61				829484	829484	15415	844899
Q-4 61				100170	105130		(10070
Q-1 62				405472	405472	0805	412278
0-3 62				2/7707	267700	5970	252111
0-4 62				541202	341202	2029	11110111
0 - 1 63				106758	106758	1785	108543
Q - 2 63				100,00			
Q-3 63				1360131	1360131	22742	1382873
Q-4 63							
0-1 64				341942	341942	7277	349219
Q-2 64				,			
Q-3 64				952629	952629	20 <b>269</b>	972898
Q-4 64							10//00
Q-1 65				189551	189921	5057	194608
0-2 65				9455441	2655761	70947	2726208
Q=3 65				2699401	2000401	10041	2120500
0-1 66				-153487	-153487	-4622	-158109
0-2 66				122101	200101		
Q-3 66				110543	110543	3330	113873
TOTAL				7414547	7414547	159806	7574353

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 9 OTHER PROGRAM ELEMENTS

.

		DESIGN /ENGR HOURS DOLLARS	PROD Hours Dollars	TOOLING AND STE HOURS DOLLARS	OTHER COST HOURS DOLLARS
DESIGN/ENGINEERING		672148	4431	591	
LABOR AT \$ 4.814		3240886	16668	2644	
ENGR BURDEN AT \$	4.682	3146367	21142	3294	
PRODUCTION			64772		
LABOR AT \$ 3.110			201412		
SHOP SUPPORT		7386			
LABOR AT \$ 3.022		22321			
TOOLING AND STE				100268	
LABOR AT \$ 3.508				351743	
PLANN ING			562	398	
LABOR AT \$ 3.203			1826	1249	
TEST/QC		653	10639	1270	
LABOR AT \$ 3.475		2005	37275	4377	
MFG BURDEN AT \$	3.650	25902	286833	366061	
ENGR MATERIAL		142558			
MEG MATERIAL			65232		
TOOLING/STE MATL				289149	
MPC		22818	6781	28950	
OTHER COST		2030233	291590	14383	5118952
SUB-TOTAL		8633090	928759	1061850	5118952
GEN & ADMIN		150361	18025	20091	109695
TUTAL COST		8783451	946784	1081941	5228647

TIME-PHASED COST				
DETAIL-SEE PAGE	II-748	II <b>-</b> 756	II <b>-</b> 762	II <b>-</b> 768

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## COST BREAKDOWNS B-70 AIRCRAFT STUDY

4-SYSTEM 9 OTHER PROGRAM ELEMENTS

	TOTAL HOURS DOLLARS
DESIGN/ENGINEERING	677170
LABOR AT \$ 4.814	3260198
ENGR BURDEN AT \$	4.682 3170803
PRODUCTION	64772
LABOR AT \$ 3.110	201412
SHOP SUPPORT	7386
LABOR AT \$ 3.022	22321
TOOLING AND STE	100268
LABOR AT \$ 3.508	351743
PLANN ING	960
LABOR AT \$ 3.203	3075
TEST/QC	12562
LABOR AT \$ 3.475	43657
MFG BURDEN AT \$	3.650 678796
ENGR MATERIAL	142558
MFG MATERIAL	65232
TOOLING/STE MATL	289149
мрс	58549
OTHER COST	7455158
SUB-TOTAL	15742651
GEN & ADMIN	298172
TOTAL COST	16040823

TIME-PHASED COST DETAIL - SEE PAGE II-769

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

### DESIGN/ENGINEERING 4-SYSTEM 9 OTHER PROGRAM ELEMENTS SUBD OF WORK DESIGN/ENGINEERING

## ON-SITE LABOR

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	58	43.5	7215	4.341	31319	32823	64142
Q-2	58						
Q-3	58 50	288.0	48384	4.274	206810	191708	398518
0-1	୦୦ 59	285-0	48574	4.439	215615	173861	389476
0-2	59	EGDEG	10211		CL / UL /	110001	505470
Q-3	59	238.5	41928	4.368	183137	154698	337835
Q-4	59						
0 - 1	60	213.0	36818	4.629	170436	138724	309160
Q-2	00	240.0	40267	4.708	189574	149748	330322
0-4	60	240.00	40201	<b>Te 100</b>	107714	149740	JJJJJ <u>C</u> E
Q-1	61	330.0	56426	4.804	271080	201472	472552
Q-2	61						
Q-3	61	178.0	32228	4.950	1,59521	151771	311292
Q-4	61						
Q-1	62	285.0	48757	5.174	252278	222108	474386
N-2	62	120 5	70.005	1 02 0	210700		( . 71
Q=3	62	430.5	12295	4.838	349198	311323	66/121
Q-1	63	376.5	64297	5.062	325458	342566	668024
Q-2	63						
Q-3	63	297.0	49952	4.849	242207	271751	513958
Q-4	63						
0-1	64	241.5	41119	4.903	201624	232135	433759
Q-2	64						
Q = 3	64	177.0	31081	5.496	170830	196751	367581
0-4	04	102.0	1 77 77 1 1	4 03 7	07110	101725	100175
$\psi = 1$ $\varphi = 2$	00 65	102.0	1111	4. 931	81440	101133	1071/2
0-3	65	48.0	8014	5.626	45084	64921	110005

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## TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

DESIGN/ENGINEERING 4-SYSTEM 9 OTHER PROGRAM ELEMENTS SUBD OF WORK DESIGN/ENGINEERING

#### ON-SITE LABOR

	MAN- Months	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-4 65						
Q-1 66	66.0	11494	5.196	59719	87194	146913
Q-2 66						
Q-3 66	93.0	15588	5.065	78956	115078	194034
TOTAL	3932.5	672148		3240886	3146367	6387253

#### **APRIL 1972**

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT 4-SYSTEM 9 OTHER PROGRAM ELEMENTS SUBD OF WORK DESIGN/ENGINEERING

		MAN- MON THS	LABOR	LABOR Rate	LABOR DOLLARS	BURDEN DOLLARS	LABOR + Burden \$
Q-3	<b>5</b> 8						
Q-4	<b>5</b> 8						
Q-1	59						
Q-2	59						
Q-3	59						
Q-4	59						
Q-1	60	12.0	2124	2.803	5953	8679	14632
Q-2	60						
Q-3	60	22.5	3793	3.079	11680	12042	23722
Q-4	60						server war in house and
0-1	61	9.0	1435	3.203	4596	4777	9373
Q-2	61						
Q-3	61		34	2.706	92	404	496
Q-4	61						
Q-1	62						
Q-2	62				Ň		
Q-3	62						
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	6.3						
Q-4	63						
0-1	64						
Q-2	64						
Q-3	64						
тот	AL	43.5	7386		22321	25902	48223

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

.

			TEST/QC			
4-SY 5	TEM	١	9	OTHER	PROGRAM	ELEMENTS
SUBD	OF	WORK	DESIGN/ENGINE	ERING		

## ON-SITE LABOR

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-1 Q-2	60 60	1.5	249	3.446	858		858
Q-3 Q-4	60 60	1.5	275	2.964	815		815
0 - 1 0 - 2	61 61		110	2.545	280		280
Q-3 Q-4	61 61		19	2.737	52		52
Q-1 Q-2	62 62						
Q-3 Q-4	62 62						
Q-1 Q-2	63 63						
Q-3 Q-4	63 63				χ.		
Q-1 Q-2	64 64						
Q-3	64						·

TOTAL 3.0

653

2005

2005

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM	9 OTHER	PROGRAM ELEMENTS
SUBD OF WORK	DESIGN/ENGINEERING	;

		MAN- MON THS	LABOR	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGR MATL
Q-1	<b>5</b> 8	43.5	7215	4.341	31319	32823	64142	
0-2 0-2	20 50	200 0	(070)		00/010	* ~ * <b>*</b> * ^		
0-4	20 59	200.0	46384	4.214	200810	191408	398518	
0 - 1	59	285.0	48574	4 430	215615	172861	200176	
0-2	59	20000	FICOP	~ <b>*</b> ● <b>*</b> <i>≥</i> ≯	219019	112001	J07410	
$\overline{0-3}$	59	238.5	41928	4.368	183137	154698	337835	
Q-4	59							
Q-1	60	226.5	39191	4.523	177247	147403	324650	2923
Q-2	60							
Q-3	60	264.0	44335	4.558	202069	161790	3638 <b>59</b>	7710
Q-4	60							
Q-1	61	339.0	57971	4.760	275956	206249	482205	2029
0-2	<b>61</b>	179 0	22291	4 644	150465	150175	211040	1.1.01
0-4	61	110.0	52201	4. 740	10,0000	192179	511040	4494
0-1	62	285.0	48757	5,174	252278	222108	474386	3957
0-2	62			2021		the last the last the last		
Q-3	62	430.5	72295	4.838	349798	317323	667121	18214
Q-4	62							
Q-1	63	376.5	64297	5.062	325458	342 566	66802 <b>4</b>	8205
Q-2	63							
0-3	63	297.0	49952	4.849	242207	271751	513958	23696
Q = 4	03	24 1 E	43110	6 002	201727	333136	477750	5540
0-2	04 64	241.5	41119	4.903	201024	232133	433139	2249
0-3	64	177.0	31081	5,496	170830	196751	367581	13898
Q-4	64	11100	51001	2. 120	110000	190191	J01/01	13330
$\bar{Q}-1$	65	102.0	17711	4.937	87440	101735	189175	14626
Q-2	65							
Q-3	65	48.0	8014	5.626	45084	64921	110005	14142
Q-4	65							
Q-1	66	66.0	11494	5.196	59719	87194	146913	13797

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 9 OTHER PROGRAM ELEMENTS SUBD OF WORK DESIGN/ENGINEERING

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	ENGF Mati
Q-2 66 Q-3 66	93.0	15588	5.065	78956	115078	194034	<b>931</b> {
TOTAL	3979.0	680187		3265212	3172269	6437481	142558

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 9 SUBD DE WORK DESTGNZE

SUBD OF WORK DESIGN/ENGINEERING

MPC COST TOTAL GEA	C OST 64 142
0-1 50 (7170	64142
WT1 00 04142	
Q-2 58	
Q-3 58 398518	398518
Q-4 58	
Q-1 59 389476	389476
Q=2 59	
Q=3 59 337835	337835
$W^{-1}$ 50 384 1579 329536 6278	335814
$W^{-2}$ 00 $W^{-3}$ 60 1014 16220 398903 7407	206.210
0-4 60	390210
$\Omega = 1 61$ 171 36095 520500 9672	530172
0-2 61	3 <b>3</b> 0114
Q-3 61 380 500801 817515 15193	832708
Q-4 61	
Q-1 62 311 368262 846916 14216	861132
Q-2 62	
Q-3 62 1436 337342 1024113 17190	1041303
Q-4 62	
<b>Q-1 63 808 103731 780768 13055</b>	793823
Q-2 63	
Q-3 63 2334 -305116 234872 3929	238801
Q-4 63	
Q-1 64 592 7754 447654 9526	457180
<b>Q-3 64 5056 18522 405057</b> 8618	413675
	222/25
	220433
עדב פט ח2 גגג מגני מסטטגג ומגג <b>ס</b> ינט א	1074451
0-4 65 2020 9200000 1040120 21920.	1014001
Q-1 66 2861 12575 176146 5305	181451

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TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 9 OTHER PROGRAM ELEMENTS SUBD OF WORK DESIGN/ENGINEERING

	MPC	OTHER Cost	SUB Total	GŁA	TOTAL COST
Q-2 66 Q-3 66	573	5881	209806	6319	216125
TOTAL	22818	2030233	8633090	150361	8783451

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING 4-SYSTEM 9 OTHER PROGRAM ELEMENTS SUBD OF WORK PRODUCTION

	MAN- Mon ths	LABOR HOURS	L'ABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1 60		8	2.625	21	28	49
0-2 60						
Q-3 60	4.5	751	3.073	2308	2810	5118
Q-4 60						
Q-1 61	12.0	2019	3.340	6743	7835	14578
Q-2 61						
Q-3 61	1.5	254	3.839	975	1906	2881
Q-4 61						
Q-1 62	1.5	161	3.472	559	744	1303
Q-2 62						
Q-3 62					40	40
Q-4 62						
0-1 63						
Q-2 63						
Q-3 63		72	8.347	601	404	1005
Q-4 63				~		
Q-1 64	7.5	1165	4.688	5461	7031	12492
Q-2 64						
Q-3 64		1			344	344
TOTAL	27.0	4431		16668	21142	37810

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	PRODUCTION			
4-SYSTEM	9	OTHER	PROGRAM	ELEMENTS
SUBD OF WORK	PRODUCTION			

		MAN- MON THS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN DOLL ARS	LABOR + BURDEN \$
Q-1	60	15.0	2535	3.409	8641	11495	20136
Q - 2	60 60	147 0	24680	2 075	75904	100770	176679
Q-4	60	T-L-L-P-O	24000	5+015	12074	100779	110013
Q-1	61	129.0	21979	3.123	68643	89110	157753
Q-2	61						
Q-3	6l	46.5	8500	2.963	25188	47942	73130
Q-1	62	18.0	3062	2.910	8911	17899	26810
Q-2	62						
Q-3	62					594	594
0-4	62						
$\sqrt{-1}$	63						
0-3	63	9.0	1587	4.100	6506	6180	12686
0-4	63						
Q-1	64	15.0	2502	3.133	7839	13077	20916
Q-2	64						
Q-3	64		-73	2.877	-210	-415	-625
тот	AL	379.5	64772		201412	286661	488073

NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	PLANNING	
4-SY STEM	9	OTHER PROGRAM ELEMENTS
SUBD OF WOR	K PRODUCTION	

		MAN- MONTHS	LABOR HOUR S	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	61		40	2.875	115	18	133
Q-4	61						
Q-1	62		9	3.556	32	6	38
Q-2	62						
Q-3	62					1	1
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	63	3.0	483	3.255	1572		1572
Q-4	63						
Q-1	64		30	3.500	105	145	250
Q-2	64						
Q-3	64				2	2	4
тот	AL	3.0	562		1826	172	1998

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TEST/QC4-SYSTEM9SUBD OF WORKPRODUCTION

ON-SITE LABOR

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN Doll Ars	LABOR + BURDEN \$
Q-1 Q-2	60 60	1.5	378	2.915	1102		1102
Q-3 Q-4	60 60	13.5	2300	3.499	8048		8048
Q-1 Q-2	61 61	19.5	3450	3.171	10939		10939
0-3 Q-4	61 61	13.5	2579	3.399	8767		8767
Q-1 Q-2 Q-3	62 62 62	9.0	1626	4.188	6809		6809
0-4 0-1 0-2	62 63 63						
Q-3 Q-4	63 63		13	7.462	9 <b>7</b>		97
Q-1 Q-2	64 64	1.5	284	5.229	1485		1485
Q-3	64		9	3.111	28		28
тот	AL	58.5	10639		37275		37275

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	4-SYSTEM 9 SUBD OF WORK PRODUCTION			OTHER PROG	RAM ELEMENTS	5		
		MAN- MON THS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$	MFG Matl
Q-1 Q-2	<b>6</b> 0 60	16.5	2921	3.343	9764	11523	21287	585
Q-3 Q-4	60 60	165.0	27731	3.110	86250	103589	189839	23519
Q-1 Q-2	61 61	160.5	27448	3.145	86325	96945	183270	22200
Q-3 Q-4	61 61	61.5	11373	3.081	3504 <b>5</b>	49866	84911	5808
Q-1 Q-2	62 62	28.5	4858	3.358	16311	18649	34960	1933
Q-3 Q-4	62 62					635	635	1516
Q-1 Q-2	63 63							
Q-3 Q-4	63 63	12.0	2155	4.072	8776	6584	15360	6871
Q-1 Q-2	64 64	24.0	3981	3.740	14890	20253	35143	2778
Q-3	64		-63	2.857	-180	-69	-249	22
TOT	AL	468.0	80404		257181	307 575	565156	65232

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 9 SUBD OF WORK PRODUCTION

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OTHER PROGRAM ELEMENTS

			TOTAL		OTHER	SUB		TOTAL
		SUBC	MATERIAL	MPC	COST	TOTAL	G & A	COST
0-1	60		585	77	12198	34147	650	34797
Q-2	60							
Q-3	60		23519	3093	3642	220093	4193	224286
Q - 4	60							
Q-1	61		22200	1876	1862	209208	3888	213096
Q-2	61							
Q-3	61		5808	491	10008	101218	1881	103099
0-4	61							
	62		1933	152	2478	39523	664	40187
Q-2	62		1 = 1 /	110				
0-4	0Z 40		1516	119	1809	4079	68	4147
01	02 43				1170	1170	20	1100
0-2	60				1179	11/9	20	1133
Q-2 Q-3	63		6871	677	71046	93954	1571	95525
Q-4	63		0071	0	11010		£.21 £	
0-1	64		2778	296	78103	116320	2475	118795
Q-2	64							
Q-3	64		22		65291	65064	1389	66453
Q-4	64							
Q-1	65				21735	21735	580	22315
Q-2	65							
Q-3	65				7157	7157	191	7348
Q-4	65							
0-1	66				7261	7261	219	7480
Q-2	66							
Q-3	66				7821	7821	236	8057
тот	AL		65232	6781	291590	928759	18025	946784

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

			DESIGN/	ENGI	NEERING		
4-SYST	EM		9		OTHER	PROGRAM	ELEMENTS
SUBD D	F	WORK	TOOLING	AND	STE		

# ON-SITE LABOR

	MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-3 60		19	4.105	78	71	149
Q-4 60						
Q-1 61		5	3.200	16	17	33
Q-2 61						
Q-3 61	1.5	195	3.621	706	865	1571
Q-4 61						
Q-1 62						
Q-2 62						
Q-3 62						
Q-4 62						
Q-1 63						
Q-2 63						
Q-3 63		30	3.400	102	166	268
Q-4 63						
Q-1 64	1.5	175	4.543	795	1062	1857
Q-2 64						
Q-3 64	1.5	167	5.671	947	1113	2060
TOTAL	4.5	591		2644	3294	5938

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> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

TOOLINGANDSTE4-SYSTEM9OTHERPROGRAMELEMENTSSUBDOFWORKTOOLINGANDSTE

		MAN- MON THS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLL ARS	LABOR + BURDEN \$
Q-3	59		5	2.200	11	13	24
Q-4	59						
Q-1	60	9.0	1610	3.376	5436	4612	10048
Q-2	60						
Q-3	60	145.5	24383	3.309	80687	79804	160491
Q-4	60						
Q-1	61	94.5	16098	3.262	52518	53803	106321
Q-2	61						
Q-3	61	70.5	12904	3.860	49810	48540	98350
Q-4	61						
Q-1	62	39.0	6652	3.411	22691	25061	47752
Q-2	62						
Q-3	62	9.0	1464	2.794	4091	7060	11151
Q-4	62						
Q-1	63	18.0	3083	3.299	10171	11746	21917
Q-2	63			-	00400	10005	00000
Q-3	63	56.0	11007	3.607	39698	40 325	80023
Q-4	63	<b>A</b> ( )	1/200	2 0 0	2 / <b>13</b> 11 13	(3000	122442
Q-1	64	96.0	16320	3.908	64 [ 5 ]	67892	132649
Q-2	64	<b>31</b> E	5122	2 10/	17070	20100	27462
Q-3	54	31+5	5422	3.180	11212	20190	31402
Q-4	64	7 5	1260	2 420	1222	4 500	10040
Q-1	02	1.0	1250	3.439	40.0.0	6009	10842
Q-2	07		60	1 1.67	366	207	545
9-3	00		00	4.401	200	271	202
тот	<b>AL</b>	586.5	100268		351743	365852	717595

## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	PLANNING			
4-SYSTEM	9	OTHER	PROGRAM	ELEMENTS
SUBD OF WORK	TOOLING AND	STE		

# ON-SITE LABOR

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	60		8	2.625	21		21
Q-4	60						
Q-1	61		98	2.888	283		283
Q-2	61						
Q-3	61						
Q-4	61						
Q-1	62						
Q-2	62						
Q-3	62						
Q-4	62						
Q-1	63					•	
Q-2	63						
Q-3	63	1.5	262	3.179	833	97	930
Q-4	63						
Q-1	64		26	3.615	94	94	188
Q-2	64						
Q-3	64		4	4.500	18	18	36
ro	TAL	1.5	398		1249	209	1458

 $\left( \begin{array}{c} & & \\ & & \\ & & \end{array} \right)$ 

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM			TEST/QC 9		OTHER	PROGRAM	ELEMENTS
SUBD	CF	WORK	TOOLING	AND	STE		

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	60		19	3.105	59		59
Q-2	60						
Q-3	60	1.5	296	3.247	961		961
Q-4	60						
Q-1	61	3.0	476	3.431	1633		1633
0-2	61						
Q-3	61		37	2.919	108		108
Q-4	61						
Q-1	62						
Q-2	62						
Q-3	62		4	3.250	13		13
Q-4	62						
Q-1	63		5	4.000	20		20
Q-2	63						
Q-3	63	1.5	227	3.573	811		811
Q-4	63						
Q-1	64		59	3.898	230		230
Q-2	64						
Q-3	64		41	3.805	156		156
Q-4	64						
Q-1	65		106	3.642	386		386
тот	AL	6.0	1270		4377		4377

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

## TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	4-SYSTEM 9 SUBD OF WORK TOOLING AND		OTHER PROGE	RAM ELEMENTS				
		MAN- MON THS	LABOR HOURS	LABOR Rate	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$	TOOL/STE Matl
Q-3	59		5	2.200	11	13	24	
Q-4 Q-1	59 60	9.0	1629	3.373	5495	4612	10107	405
Q-2 Q-3	60 60	147.0	24706	3.309	81747	<b>7</b> 9875	161622	42374
Q-4 Q-1	60 61	97.5	16677	3.265	5445C	53820	108270	35983
Q-2 Q-3	61 61	72.0	13136	3.854	50624	49405	100029	72363
Q-4 Q-1	61 62	39.0	6652	3.411	22691	25061	47752	56015
Q-2 Q-3	62 62	9.0	1468	2.796	4104	7060	11164	41095
Q-4 Q-1 Q-2	62 63	18.0	3088	3.300	10191	11746	21937	
Q-3	63 63	69.0	11526	3.596	41444	40588	82032	1548
Q-4 Q-1	64 64	97.5	16580	3.973	65876	69048	134924	30866
Q-3	64 64	33.0	5634	3.265	18393	21321	39714	8210
Q-4 Q-1	64 65	7.5	1366	3.455	4719	6509	11228	290
Q-2 Q-3	65 65		60	4.467	268	297	565	
тот	AL	598.5	102527		360013	369355	729368	289149

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NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 9 OTHER PROGRAM ELEMENTS SUBD OF WORK TOOLING AND STE

	MPC	OTHER COST	SUB TOTAL	GεA	TOTAL COST
Q-3 59			24		24
Q-4 59 Q-1 60	53	2224	12789	243	13032
Q-2 60					
Q-3 60 Q-4 60	5572	46	209614	3994	213608
9 - 1 61	3041		147294	2737	150031
0-2 61			2	2121	100001
Q-3 61	6115	1162	179669	3339	183008
Q=4 01 0-1 42	4414		109101	1014	100007
$Q = 1 \ 62$	4414		100101	1910	103331
Q-3 62	3238		5549 <b>7</b>	932	56429
Q-4 62					
Q-1 63			21937	367	22304
Q-2 63	150		00 70 0	1 ( 0.0	0 - 1
Q = 3  63	152		83132	1400	85132
Q = 4 0.5	2200	2270	170050	2//0	17(007
Q = 1 04	5290	5219	112000	2006	110021
Q-3 64	298 <b>7</b>	4987	55898	1189	57087
Q-4 64					
Q-1 65	88		11606	310	11916
Q-2 65					
Q-3 65			565	15	580
Q-4 65		1070	1070	<b>.</b>	1100
Q-1 00		1070	1070	52	1102
Q-3 66		1615	1615	49	1664
TOTAL	28950	14383	1061850	20091	1081941

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# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

# TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	4-SYSTEM Subd of	9 WORK OTH	IER COST	OTHER PROGE	AM ELEMENTS	
		SUBC	OTHER COST	SUB TOTAL	G&A	TOTAL COST
Q-1	60		18116	18116	345	18461
Q-2	60		110000	110000		
0	60 60		112353	112353	2141	114494
0-1	61		66498	66498	1236	67734
$\bar{Q}-2$	61		00170	00170	1200	01104
Q-3	61		318024	318024	5910	323934
Q-4	61					
Q-1	62		35864	35864	602	36466
0-2	62					
Q-3	62		11976	11976	201	12177
Q-4 0-1	63		5697	5607	0.5	6707
0 - 2	63		2071	2091	30	2192
Q-3	63		1598681	1598681	26730	1625411
Q-4	63					
Q-1	64		256870	256870	5466	262336
Q-2	64					
Q-3	64 4 /		869076	869076	18492	887568
0-4 0-1	04 65		163492	163492	4362	167854
<b>3</b> -2	65		100472	103472	4002	101074
<b>a</b> -3	65		1730851	1730851	46179	1777030
Q-4	65					
<u>9-1</u>	66		-168921	-168921	-5087	-174008
2-2	66					
1-3	65		100375	100375	3023	103398
тот	AL		5118952	5118952	109695	5228647

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

design/engineering4-system9Other program elements

# ON-SITE LABOR

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLAR S	BURDEN Doll Ars	LABOR + BURDEN \$
Q-1	58	43.5	7215	4.341	31319	32823	64142
Q-2 Q-3	58 58	288.0	48384	4.274	206810	191708	398518
Q-1 Q-2	59 59	285.0	48574	4.439	215615	173861	389476
Q-3 Q-4	59 59	238.5	41928	4.368	183137	154698	337835
Q-1 Q-2	60 60	213.0	36826	4.629	170457	138752	309209
ର−3 Q−4	60 60	244.5	41037	4.678	191960	152629	344589
Q-1 Q-2	61 61	342.0	58450	4.753	277839	209324	487163
Q-3 Q-4	61 61	180.0	32677	4.933	161202	154542	315744
Q-1 Q-2	62 62	286.5	48918	5.169	252837	222852	475689
Q-3 Q-4	62 62	430.5	72295	4 • 83 8	<b>3497</b> 98	317363	667161
Q-1 Q-2	63 63	376.5	64297	5.062	325458	342566	668024
Q-3 Q-4	63 63	298.0	50054	4.853	242910	272321	515231
Q-1 Q-2	64 64	249.0	42459	4.896	207880	240228	448108
Q-3 Q-4	64 64	177.0	31249	5.497	171777	198208	369985
Q-1 Q-2	65 65	102.0	17711	4.937	87440	101735	189175
Q-3 Q-4	65 65	48.0	8014	5.626	45084	64921	110005

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

#### DESIGN/ENGINEERING 9 OTHER PRO

# 4-SYSTEM

OTHER PROGRAM ELEMENTS

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL AR S	LABOR + BURDEN \$
Q-1 66	66.0	11494	5.196	59719	87194	146913
Q-3 66	93.0	15588	5.065	78956	115078	194034
TOTAL	3961.0	677170		3260198	3170803	6431001

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM

PRODUCTION 9

OTHER PROGRAM ELEMENTS

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-1	60	15.0	2535	3.409	8641	11495	20136
Q-2	60						
Q-3	60	147.0	24680	3.075	75894	100779	176673
Q-4	60					,	
Q-1	61	129.0	21979	3.123	68643	89110	157753
Q-2	61						
Q-3	61	46.5	8500	2.963	25188	47942	73130
Q-4	61						
Q-1	62	18.0	3062	2.910	8911	17899	26810
Q-2	62						
Q-3	62					594	594
Q - 4	62						
Q-1	63						
Q-2	63						
Q-3	63	9.0	1587	4.100	6506	6180	12686
Q - 4	63						
Q-1	64	15.0	2502	3.133	7839	13077	20916
0-2	64						
0-3	64		-73	2.877	-210	-415	-625
тот	AL	379.5	64772		201412	286661	488073

#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100 ,

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

SHOP SUPPORT

# 4-SYSTEM

#### 9

OTHER PROGRAM ELEMENTS

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN DOLLARS	LABOR + BURDEN \$
Q-3	<b>5</b> 8						
Q-4	58						
Q-1	59						
Q-2	59						
Q-3	59						
Q-4	59						
Q-1	60	12.0	2124	2.803	5953	8679	14632
Q-2	60						
Q-3	60	22.5	3793	3.079	11680	12042	23722
Q-4	60						
Q-1	61	9.0	1435	3.203	4596	4777	9373
Q-2	61						
Q-3	61		34	2.706	92	404	496
Q-4	61						
	62						
Q-2	62						
0-4	02 47						
0-1	62					•	
0-2	63						
0-3	63						
Q-4	63						
0-1	64						
$\hat{\mathbf{Q}} - \hat{\mathbf{Z}}$	64						
Q-3	64						
TOT	TAL	43.5	7386		22321	25902	48223

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

**TOOLING AND STE4-SYSTEM**9OTHER PROGRAM ELEMENTS

# ON-SITE LABOR

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN Dollars	LABOR + BURDEN \$
Q-3	59		5	2.200	11	13	24
Q-4	59						
Q-1	60	9.0	1610	3.376	5436	4612	10048
Q-2	60						
Q-3	60	145.5	24383	3.309	80687	79804	160491
Q-4	60						
Q-1	61	94.5	16098	3.262	52518	53803	106321
Q-2	61						
Q-3	61	70.5	12904	3.860	49810	48540	98350
Q-4	61						
0-1	62	39.0	6652	3.411	22691	25061	47752
Q-2	62						
Q-3	62	9.0	1464	2.794	4091	7060	11151
Q-4	62						
Q <b>-1</b>	63	18.0	3083	3.299	10171	11746	21917
Q-2	63						
Q-3	63	66.0	11007	3.607	39698	40325	80023
Q-4	63						
Q-1	64	96.0	16320	3.968	64757	67892	132649
Q-2	64		- (	- 107		20100	
Q-3	64	31.5	5422	3.186	17272	20190	37462
Q-4	64				(	( 500	100/0
Q-1	65	7.5	1260	3.439	4333	6509	10842
Q-2	65				24.0	00 <b>7</b>	5 / <b>5</b>
Q-3	65		60	4.467	268	297	56.5
тот	T AL	586.5	100268		351743	365852	717595

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#### NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

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#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

PLANNING

# 4-SYSTEM

OTHER PROGRAM ELEMENTS

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLAR S	BUR DEN DOLL ARS	LABOR + BURDEN \$
Q-3	60		8	2.625	21		21
Q-4	60						
Q-1	61		98	2.888	283		283
Q-2	61						
Q-3	61		40	2.875	115	18	133
Q-4	61						
Q-1	62		9	3.556	32	6	38
Q-2	62						
Q-3	62					1	1
Q-4	62						
Q-1	63						
Q-2	63						
Q-3	63	4.5	745	3.228	2405	9 <b>7</b>	2502
Q-4	63						
Q-1	64		56	3.554	199	239	438
Q-2	64						
Q-3	64		4	5.000	20	20	40
тот	TAL	4.5	960		3075	381	3456

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM

TEST/QC 9

OTHER PROGRAM ELEMENTS

		MAN- MONTHS	LABOR HOURS	LABOR RATE	LABCR DOLLARS	BUR DEN DOLLARS	LABOR + BURDEN \$
Q-1	60	4.0	646	3,125	2019		2019
Q=2 Q=3	60 60	16.5	2871	3.422	9824		9824
Q-1 Q-1	61 61	24.0	4036	3.184	12852		12852
Q-2 Q-3	61	15.0	2635	3.388	8927		8927
Q-4 Q-1	61 62	9.0	1626	4.188	6809		6809
Q-2 Q-3	62 62		4	3.250	13		13
Q-1	63		5	4.000	20		20
Q-2 Q-3	6.3 63	1.5	240	3.783	908		908
Q-4 Q-1	63 64	1.5	343	5.000	1715		1715
Q-2 Q-3	64 64		50	3.680	184		184
Q-4 Q-1	64 65		106	3.642	386		386
тот	AL	71.5	12562		43657		43657

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## NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

		MAN- MONTHS	LABOR HOUR S	LABOR RATE	LABOR DOLLARS	BURDEN Doll Ars	LABOR + BURDEN \$	ENGR Matl
Q-1	58	43.5	7215	4.341	31319	32823	64142	
Q-2	58							
Q-3	58	288.0	48384	4.274	206810	191708	398518	
Q-4	58							
Q-1	59	285.0	48574	4.439	215615	173861	389476	
Q-2	59							
Q-3	59	238.5	41933	4.368	183148	154711	337859	
Q-4	59	050 0						
9-1 0 0	60 70	253.0	43/41	4.401	192506	163538	356044	2923
Q-2	60	57( 0	01770	7 004	2700//	- / F - F /	315363	
0-1	60	270.0	90112	3.824	370066	345254	115320	1110
0-1	61	509 5	102006	6 092	616721	257014	777765	30.30
0-2	61	د ۲۰۰۰	102095	4.002	410/51	557014	(15145	2029
0-3	61	312.0	56790	4 320	745334	251446	496780	4494
Q-4	61	212.00	20170	70 320	243334	231440	4 30 100	4474
0 - 1	62	352.5	60267	4.833	291280	265818	557098	3957
0-2	62					209020	221.020	
0-3	62	439.5	73763	4,798	353902	325018	678920	18214
Q-4	62							
Q-1	63	394.5	67385	4.981	335649	354312	689961	8205
Q-2	63							
Q-3	63	379.0	63633	4.596	292427	318923	611350	23696
Q-4	63							
Q-1	64	361.5	61680	4.578	282390	321 436	603826	5549
Q-2	64							
Q-3	64	208.5	36652	5.158	189043	218003	407046	13898
Q-4	64							
Q-1	65	109.5	19077	4.831	92159	108244	200403	14626
Q-2	65							•
Q-3	65	48.0	8074	5.617	45352	65218	110570	14142
Q-4	65							
Q-1	66	66.0	11494	5.196	59719	87194	146913	13797

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	MAN- MONTHS	LABOR HOURS	LABOR RATE	LABOR DOLLARS	BURDEN Doll Ars	LABOR + BURDEN \$	ENGR MATL
0-2 66 0-3 66	93.0	15588	5.065	78956	115078	194034	9318
TOTAL	5046.5	863118		3882406	3849599	7732005	142558

# NORTH AMERICAN ROCKWELL CORP. SPACE DIVISION DATA PREPARED UNDER NASA CONTRACT NAS9-12100

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

		MFG	TCOL/STE		TOTAL		OTHER	SUB
		MATL	MATL	SUBC MA	TERIAL	MPC	COST	TOTAL
Q-1	58							64142
Q-2	58							
Q-3	58							398518
Q-4	58							
Q-1	59							389476
Q-2	59							
Q-3	59							337859
Q-4	59							
Q = 1	60	585	405		3913	514	34117	394588
0-2	60 60	22510	10071		77.000	0/70	1 2 2 2 4 1	
0-4	60	2.3313	42.574		13603	9019	132261	930863
Q-1	61	22200	35983		60212	5088	104455	942500
Q-2	61		02700		00212	2000	107775	949900
Q-3	61	5808	72363		82665	6986	829995	1416426
Q-4	61							
Q-1	62	1933	56015		61905	4877	406604	1030484
Q-2	62							
Q-3	62	1516	41095		60825	4793	351127	1095665
0-4	62				0.005	202		
0-2	62				8205	808	110607	809281
0-3	63	6871	1548		32115	3163	1364611	2011239
0-4	63		1210		26112	5105	1904011	2011237
Q-1	64	2778	30866		39193	4178	346006	993203
Q-2	64							
Q-3	64	22	8210		22130	8043	957876	1395095
Q-4	64							
Q-1	65		290		14916	4463	191759	411541
Q-2	65							,
Q-3	65 45				14142	2523	2658063	2785298
Q−4 ∩_1	07 66				12707	30/1	-149015	16667
~- I	00				12121	2001	-140010	10000

#### TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

	MEG . MATL	TCOL/STE MATL	SUBC	TOTAL MATERIAL	MPC	OT HE R COS T	SUB TOTAL
Q-2 66 Q-3 66				9318	573	115692	319617
TOTAL	65232	289149		496939	58549	7455158	15742651

#### TIME PHASED EXPEND. 8-70 AIRCRAFT STUDY

4-SYSTEM 9 UTHER PROGRAM ELEMENTS

		GεA	TOTAL COST
0-1	58		64142
Q-2	58		
Q=3	58 50		398518
0 - 1	20 50		700176
0-2	59		209410
0-3	59		337859
$\overline{Q-4}$	59		551077
Q-1	60	7516	402104
Q-2	60		
Q = 3	60	17735	<b>54</b> 8598
Q-4	60		
Q-1	61	17533	961033
Q-2	51		
Q-3	61	26323	1442749
Q-4	61		
Q-1	62	17298	1047782
Q-2	62	10201	111/05/
0-4	02 62	10 291	1114020
u = 4 0 = 1	62	12527	822118
ũ-2	63	1 6 6 6 A	020110
Q-3	63	33630	2044869
Q-4	63		
Q-1	64	21135	1014338
0-2	64		
Q-3	64	29688	1424783
Q-4	64		
Q-1	65	10979	422520
u-2	55 75	77/ 211	2050/00
Q-3	07	14311	2824004
N-4 0-1	66	459	16025
ur L	<b></b>	- U -	

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> TIME PHASED EXPEND. B-70 AIRCRAFT STUDY

4-SYSTEM 9 OTHER, PROGRAM ELEMENTS

	G & A	TOTAL Cost
Q-2 66 Q-3 66	9627	329244
TOTAL	298172	16040823

**APRIL 1972**