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Off-Design Computer Code for Calculating the Aerodynamic Performance of Axial-Flow Fans and Compressors

User's Manual

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(NASA-CR-198362) OFF-DESIGN
COMPUTER CODE FOR CALCULATING THE
AERODYNAMIC PERFORMANCE OF
AXIAL-FLOW FANS AND COMPRESSORS
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SUMMARY

An off-design axial-flow compressor code is presented and is available from COSMIC for predicting the aerodynamic performance maps of fans and compressors. Steady axisymmetric flow is assumed and the aerodynamic solution reduces to solving the two-dimensional flow field in the meridional plane. A streamline curvature method is used for calculating this flow-field outside the blade rows. This code allows for bleed flows and the first five stators can be reset for each rotational speed; capabilities which are necessary for large multi-stage compressors. The accuracy of the off-design performance predictions depend upon the validity of the flow loss and deviation correlation models. These empirical correlations for the flow loss and deviation are used to model the real flow effects and the off-design code will compute through small reverse flow regions.

The input to this off-design code is fully described and a user's example case for a two-stage fan is included with complete input and output data sets. Also, a comparison of the off-design code predictions with experimental data is included which generally shows good agreement.

1.0 INTRODUCTION

If a reliable off-design axial compressor code is available for predicting the performance maps of compressor designs, costly and time consuming testing can be kept to a minimum. In addition, reliable flow property prediction at off-design speeds can be used as a good starting point for full three dimensional flow analyses to reduce their long computing times. Off-design compressor performance, which is the performance of a compressor at flow conditions and rotational speeds other than the design point is an important aspect of compressor design. A good design for a multistage axial compressor requires an adequate part-speed performance with sufficient part-speed stall margin and determination of required starting bleed flows along with any guide vane or stator resets needed to better match the stages (Ref. 1). The off-design flow analysis is different from the compressor design analysis in that the compressor blade geometry is now fixed and the off-design requirement is to determine the compressor performance for a range of speeds and weight flows i.e. to determine the compressor map.

This off-design, two-dimensional, axial compressor code was originally developed by James Crouse as an analysis code to be used in tandem with his compressor design code (Ref. 2). This off-design code has never been published and many modifications to the original code have been made over the last 12 years. In order to successfully use this off-design code, the compressor design code must first be used to generate the necessary blade geometry block of data that is part of the required input to the off-design code. Actually, the general design procedure is to first produce a preliminary initial blade design with the compressor design code. In cases where the compressor blade design is obtained from an old design or an outside source, the compressor design code must be used to duplicate the blade design and obtain the necessary input to the off-design code.

Steady, axisymmetric, two-dimensional flow in the meridional plane is as-

sumed and the equations of motion as developed in reference 1 are only applicable for calculation stations outside the blade rows. Also, empirical correlations for the flow losses and deviation angle are used to model real flow effects and the off-design code will compute through small reverse flow regions.

The blade geometry description is essentially the same as that for the compressor design code (Ref. 2) and the aerodynamic solution uses a streamline curvature method for calculating the flow at stations outside the blade rows. The aerodynamic calculation structure for the off-design code is set up such that the aerodynamic flow field and selected streamlines are recalculated for each new rotational speed with the flow rate increased to the choke condition, and then decreased to the stall condition. In addition, this code has the capability to include bleed flows as well as reset adjustments for the first five stators.

The purpose of this report was not intended to give a detailed description of the development of the off-design code but to give the user a manual for properly using this code.

A user's example case for a two-stage fan is included to demonstrate the necessary input required to run this off-design code as well as the resulting output produced by this code. A comparison of the off-design code performance predictions with experimental data for this two-stage fan is made with and without stator resets.

2.0 OFF-DESIGN CALCULATION PROCEDURE

Similar to the compressor design code (Ref. 2) the off-design code is separated into three main regions of calculation: the input and setup, the aerodynamic flow solution and exit calculation routines. In addition, a reset or recalculation routine is included for a recalculation of the flow field and streamlines as the flow is increased toward the choke point and decreased toward the stall condition. This general procedure is repeated for each desired rotational speed.

For the input and setup routine the input data are read and interpreted and the streamlines are located on the basis of annulus area. For the flow field solution phase the equations of motion are satisfied in the meridional ($r-z$) plane at calculation stations along the compressor flow path (see Figure 1). At the stations, the equations of motion and overall flow continuity are satisfied using a streamline curvature method of solution. The exit calculation routine makes the final calculations and prints the output. First the mass averaged parameters for the individual and cumulative compressor blade rows are calculated and printed and then the detailed tabulated local values (tip to hub) of the aerodynamic and blading parameters for each calculation stations are computed and printed. A recalculation phase or controlling main routine is used to recalculate the flow field and streamlines as the inlet flow rate is increased toward the choke point and decreased toward the stall condition. This procedure is repeated for each desired compressor rotational speed.

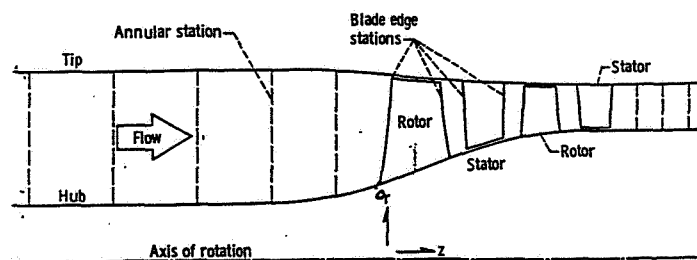


Figure 1 - Calculation stations in compressor flow path.

3.0 INPUT DATA

The basic meridional (r, z) computational plane is shown in Figure 1 for a typical compressor configuration. As indicated in Figure 1 the calculation stations are located at the blade-row leading and trailing edges and at other annular locations needed for proper determination of the streamlines. The input data are divided into two categories of input: general input data and calculation station and blade row input data. All the input parameters and options as well as the input data format are described in Appendix B. The mathematical symbols used for defining these input parameters are given in Appendix A. Except for a few modifications the general input data for the off-design code are similar to the general input data for the design code (Ref. 2).

3.1 General Input Data

The input data consist of the following information:

- 1). Number of desired compressor speeds.
- 2). Normalized range for the choke region.
- 3). Normalized step increment for the choke region.
- 4). Normalized range for the stall region.
- 5). Normalized step increment for the stall region.
- 6). Fraction of desired rotation speeds.
- 7). Stator reset angle for first 5 stators.
- 8). Number of streamlines.
- 9). Number of blade rows.
- 10). Number of annular stations.
- 11). Compressor rotational speed.
- 12). Inlet flow rate.
- 13). Desired compressor pressure ratio.
- 14). Gas molecular weight.
- 15). Coefficients for C_p as a fifth-degree polynomial function of temperature.

- 16). Upstream values of total temperature, total pressure, and inlet tangential velocity for each streamline.
- 17). Streamtube mass flow fractions between streamlines.
- 18). Sets of points to define tip and hub casing contours.
- 19). Sets of blade element profile loss parameters that are tabulated as functions of blade element loading parameter (D Factor) and fraction of passage height.

Except for the above first seven input data items the remaining list of the general input data is identical to the general input data for the design code (Ref. 2). Therefore, once the design code is run most of the general input data for the off-design code are already known.

3.2 Calculation Station and Blade Row Input

The input data for the annulus calculation stations and blade rows are read in order from annulus inlet to outlet. The calculation stations and blade row input consist of the following information:

I. Type of calculation station: annular, rotor or stator.

If annular - only one line.

- 1). The tip and hub axial locations.
- 2). The tip and hub blockage displacement thickness.
- 3). The cumulative bleed fraction of inlet flow

If rotor - first line.

- 1). The tip and hub stacking-point axial locations.
- 2). The inlet tip and hub blockage displacement thicknesses.
- 3). The cumulative bleed fraction of inlet flow.

If rotor - second line.

- 1). Number of streamlines (maximum of 11).
- 2). Loss set number (maximum of 9).

- 3). Blade definition index.
- 4). Option parameter for part span shroud ($IDAMP = 0$ for no damper).
- 5). The cumulative bleed factor of inlet flow.
- 6). Number of blades.
- 7). The outlet tip and hub blockage displacement thicknesses.
- 8). The angle of stacking axis tilt (degrees) or lean in circumferential direction (r, θ plane).
- 9). The estimated cumulative stage adiabatic efficiency for the first iteration.
- 10). Desired cumulative energy addition factor through particular rotor to total energy addition of compressor. The last rotor must have $CRENGY = 1.0$ to meet the input total pressure ratio.

If stator - first line.

- 1). The tip and hub stacking-point axial location.
- 2). The inlet tip and hub blockage displacement thicknesses.
- 3). The cumulative bleed fraction of inlet flow.

If stator - second line.

- 1). Number of streamlines (maximum of 11).
- 2). Loss set number (maximum of 9).
- 3). Blade definition index.
- 4). Option parameter for part span shroud ($IDAMP = 0$ for no damper).
- 5). The cumulative bleed fraction of inlet flow.
- 6). Number of blades.
- 7). The outlet tip and hub blockage displacement thicknesses.

If $IDAMP > 0$, then the following information line needs to be included after the second line of the rotor or stator input.

- 1). Radial location of the leading edge or the part-span shroud.
- 2). Slope angle of the damper centerline in the meridional view.

- 3). Chord length of mid-span shroud.
- 4). Maximum thickness-to-chord ratio of damper.
- 5). Leading edge to chord ratio of a part-span shroud.
- 6). Trailing edge to chord ratio of a part-span shroud.

The basic blade row geometry block of information is included after the second line of the rotor or stator blade row input. This basic blade row geometry data is automatically obtained using an option in the compressor design code (Ref. 1). Again, as previously stated, the order and format for all of the off-design input is shown in Appendix B along with a complete description of the input parameters.

4.0 AERODYNAMIC FLOW SOLUTION

The aerodynamic flow solution determines the velocity diagrams and fluid state properties on the calculated streamlines at the annular and blade row inlet and exit stations. The flow solution results from an inner and outer iteration scheme. For the outer loop iteration the calculated variables are the stagnation temperature and pressure, the tangential component of velocity, and the streamline location, slope, and curvature. The inner loop iteration is the spanwise flow calculation, solving continuity and radial equilibrium at each station, where the axial velocity distribution is calculated as the outer loop variables are held fixed and then the streamline location is reset. The overall procedure is station marching from inlet to outlet with streamline parameters fixed and a maximum of 20 cycles to converge to a solution.

4.1 Flow Loss and Deviation Prediction Models

Accuracy of the off-design performance predictions depends upon the validity of the flow loss and deviation correlation models. The total loss coefficient is defined as:

$$\bar{w} = \frac{(PO'_i)_2 - PO'_2}{PO'_1 - p_1} \quad (1)$$

In this code the losses consist of two additive components: shock losses and all other losses (considered as a profile loss). The shock loss is assumed to be approximately equal to the computed normal shock loss divided by the square of the inlet relative Mach number. The profile loss coefficient is given by:

$$\bar{w}_p = \bar{w} - \bar{w}_s \quad (2)$$

where $\bar{w}_s$ is the shock loss coefficient.

The profile loss is represented by a correlation parameter as a function of passage height and aerodynamic blade loading. The values for this loss parameter correlation are input in tabular form in the input data and are usually called

loss parameter data sets. The aerodynamic blade loading parameter (more commonly called D-factor) is defined as:

$$D = 1 - \frac{V'_2}{V'_1} + \frac{[(rV_\theta)_2 - (rV_\theta)_1]}{\sigma(r_1 + r_2)V'_1} \quad (3)$$

where V_θ is the tangential flow velocity and σ is the blade element solidity and the loss parameter in the input table is defined as:

$$\frac{\bar{w}_p \cos \beta'_2}{2\sigma} \quad (4)$$

Loss parameter data sets determined from experimental data over many years are often stored in data banks for different types of compressors.

In this off-design code a deviation subroutine is used to iterate for the blade row outlet flow angle and thus with fixed blade geometry determines the deviation angle. A correction to Carter's deviation angle rule (Ref. 1) due to changes in blade camber in the streamline direction was developed by James Crouse in his original development of the off-design code. This correlation for the deviation coefficient used in the iteration for the outlet blade flow angle is given by:

$$\begin{aligned} C_m = & (0.219 + \gamma(0.051085 + 0.088915\gamma)) \\ & \times \{1.620 + \gamma(0.72365 - 0.27871\gamma) \\ & + \phi[-1.415 + \gamma(1.85240 - 0.75964\gamma) \\ & + \phi(0.352 - \gamma(0.810816 - 0.40444\gamma))]\}^{2.4} \end{aligned}$$

where: γ is the blade setting angle (rad.), c is the chord, and ϕ is the ratio of front to total camber of each blade element.

These loss and deviation coefficient correlations are the primary unknown factors in the off-design code. Generally, the off-design code predictions compare quite well with experimental data for fans and lightly to moderately loaded

multi-stage compressors. However, for highly-loaded, large multistage compressors very large reset guide vane and stator adjustments are needed to match the stages, especially at low wheel speeds. These large reset adjustments greatly increase the pressure losses and deviation angles in these front stators and guide vane. Therefore, the loss and deviation correlations need to be improved for highly-loaded, large, multistage compressors.

5.0 EXIT CALCULATION ROUTINE

After the iterated solution for the aerodynamic flow field is obtained the off-design code calculates and prints the overall mass-averaged parameters for the individual and cumulative compressor blade rows. In addition, the detailed tabulated local values (tip to hub) of the aerodynamic and blading parameters for each calculation station are computed and printed.

6.0 CONTROLLING RECALCULATION MAIN ROUTINE

The first flow field and performance calculation for this off-design code is the design point case and the results should compare very well with the results for the design code (Ref. 2). Then, the step-choke increment is used to calculate an increased flow rate and the flow field with new streamlines are recalculated. This process is repeated until the input range-choke value is reached and all the performance calculations on the choke side are completed. A word of caution is needed at this point because if too large a value for the input range-choke is used, the code will try to calculate a very hard choke condition which may cause a divide by almost zero and other numerical difficulties. After the choke-side calculation is completed, the off-design code then uses the step-stall increment to reduce the flow rate from the design value and the flow field with new streamlines are calculated. This process on the stall side is repeated until the desired input range-stall value is reached. Again, caution is needed because if too large a value for the input range-stall is used, the code will try to calculate a stall condition with very high D-factors (Eq. 3) resulting in severe numerical difficulties which can stop the code. Now, all the flow-field overall and blade row performance calculations have been obtained for one complete speedline (100 percent or fraction of speed of 1.0). This same general procedure is repeated for additional speedlines except that instead of marching away from design point values the controlling main routine calculates a reduced flow rate and rotational speed for each additional desired input fraction of speed. Speedlines greater than 100 percent or fraction of speed greater than 1.0 can also be calculated.

7.0 USER EXAMPLE COMPRESSOR CASE

A user's example case for a two-stage fan (Ref. 3) shows the necessary detailed input data set needed to run this off-design code (see Table 1). This two-stage fan example case is the same example case that was used for the compressor design code (Ref. 2). In addition, the off-design output prediction results for this example case with and without stator resets are shown in Table 2 and Table 3, respectively. A comparison of the off-design predictions is generally in good agreement with the experimental data (see Fig. 2), especially for the 70 percent speed case.

This compressor off-design computer code (written in Fortran) along with the example input case can be obtained from COSMIC, 112 Barrow Hall, University of Georgia, 30601. The COSMIC program number is LEW-16176.

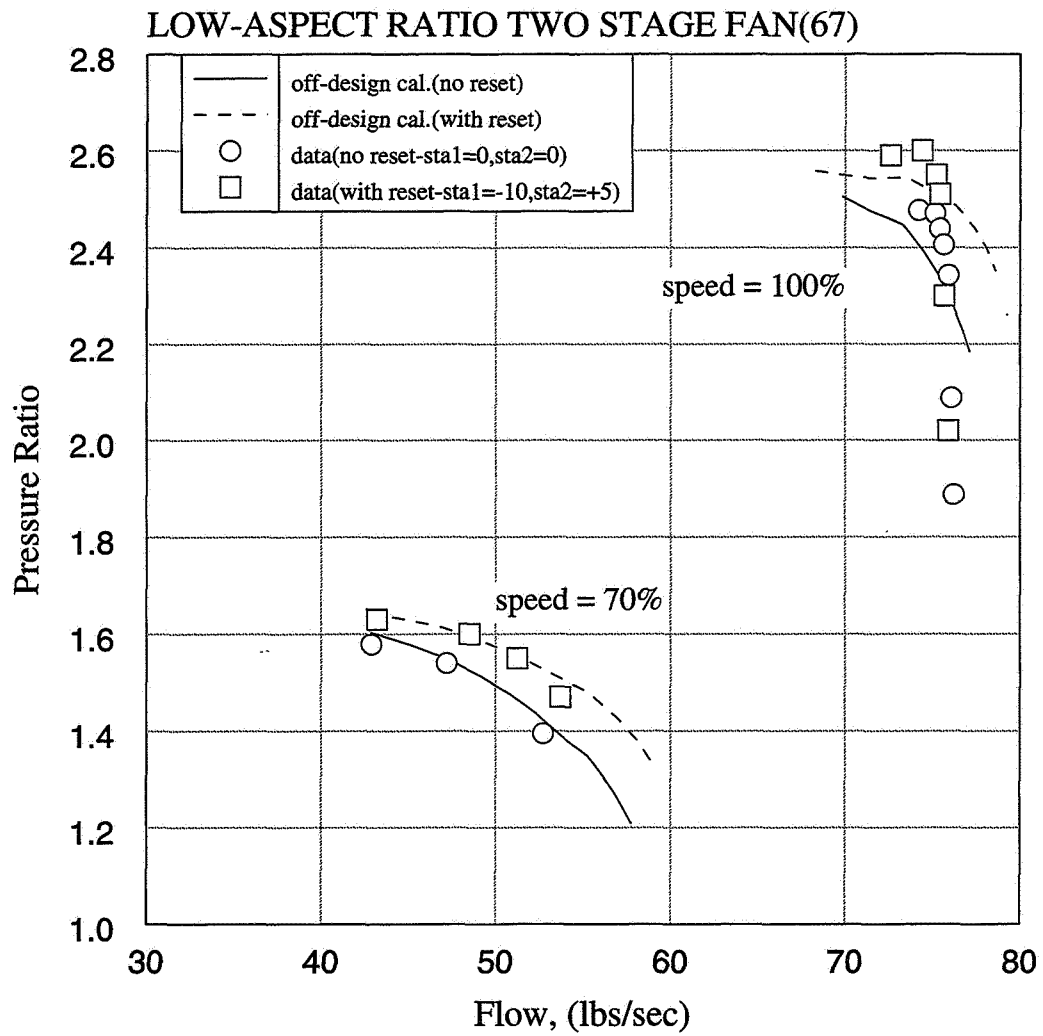


Figure 2 - Comparison of the off-design code performance predictions with experimental data for a low-aspect-ratio two-stage fan with and without stator reset.

8.0 ACKNOWLEDGEMENT

The original version of this off-design code was developed by James Crouse while he was an employee of NASA. Even after he left NASA James Crouse has provided several corrections and additions as well as many suggestions to improve this code and his valuable help is greatly appreciated.

9.0 REFERENCES

1. Johnson, I.A., and Bullock, R.O. eds., "Aerodynamic Design of Axial-Flow Compressors", NASA SP-36, 1965.
2. Crouse, J.E., "Computer Program for Aerodynamic and Blading Design of Multi- Stage Axial-Flow Compressors", NASA TP-1946, 1981.
3. Urasek, D.C., Gorrell, W.T., and Cunnam, W.S., "Performance of Two-Stage Fan Having Low-Aspect Ratio, First-Stage Rotor Blading", NASA TP-1493, 1979.

APPENDIX A

Symbols

A	annulus area, ft^2	α	angle of streamline with reference to axial direction, deg.
c	blade chord, in.	β	flow angle relative to meridional direction, deg.
$C_p(T)$	specific heat function for constant pressure, $ft/sec^2 \text{ } ^\circ R$	γ	blade setting angle, deg.
C_m	empirical correlation for the deviation angle coefficient	δ	deviation angle, deg.
D	blade element diffusion factor	θ	circumferential direction, rad
I	integer station index	ϕ	ratio of front to total camber
i	incidence angle, deg.	κ	blade angle relative to local conic co-ordinate ray, deg.
J	integer streamline index	ρ	local density, slugs/ ft^3
K	ineger loss set index	σ	local blade solidity, ratio of chord to circumferential spacing
k	curvature in curvilinear co-ordinate system	$\bar{\omega}$	total loss coefficient
L	distance along chord line, in.	$\bar{\omega}_s$	shock loss coefficient
M	Mach number	$\bar{\omega}_p$	profile loss coefficient
m	streamline direction in meridional plane, in.		
p	static pressure, lb/ft^2		
PO	stagnation pressure, lb/ft^2 ; input values; lb/in^2	Subscripts:	
$\mathcal{R}$	gas constant for air, $ft\text{-}lb/slug \text{ } ^\circ R$	I	calculation station index
r	radius from axis of rotation, in.	i	ideal value, as by an isentropic process
S	blade element path distance, in.	le	leading edge
TO	stagnation temperature, $^\circ R$	m	streamline direction in meridional plane; also, maximum blade thickness
t	static temperature, $^\circ R$; also, blade element thickness, in.	o	initial value
U	local blade velocity, ft/sec .	t	transition point
V	velocity, ft/sec .	te	trailing edge
W	weight flow, ft/sec .	θ	circumferential direction
z	axial distance, in.	1	blade row inlet
		2	blade row outlet
		Superscripts:	
		$()'$	relative to rotor

The input variables for the off-design compressor code and various options are described in this Appendix. The calculation station and blade row data sets are input in the order in which they occur in the compressor. The format for the input data is given in the following block diagrams.

18

		1		2		3		4		5		6		7		8	
1	2	3	4	5	6	7	8	9	0	1	2	3	4	5	6	7	8
Option line: Read when IDAMP > 0 (blade row damper geometry)																	
RDAMP(IROW)		ADAMP(IROW)		CDAMP(IROW)		TDAMP(IROW)		RDLE(IROW)		RDTE(IROW)							
Basic blade row geometry . . . All the following blade row geometry input is																	
automatically obtained using an option in the compressor design code (NASA TP-1946).																	
THLE(1)		THPE(1)		THMAX(1)		ZM(1)		T(1)		P(1)							
KTC(1)		KOC(1)		CHORD(1)		ALP(1)		RSP(IROW,1)		ZS(1)		TS(1)					
THLE(2)		THPE(2)		THMAX(2)		ZM(2)		T(2)		P(2)							
KTC(2)		KOC(2)		CHORD(2)		ALP(2)		RSP(IROW,2)		ZS(2)		TS(2)					
THLE(NEL)		THPE(NEL)		THMAX(NEL)		ZM(NEL)		T(NEL)		P(NEL)							
KTC(NEL)		KOC(NEL)		CHORD(NEL)		ALP(NEL)		RSP(IROW,NEL)		ZS(NEL)		TS(NEL)					
Option: When IGEO > 0 (additional blading geometry)																	
ACF(IROW,1)		BCF(IROW,1)		CCF(IROW,1)		DCF(IROW,1)		ACR(IROW,1)		BCR(IROW,1)		CCR(IROW,1)		DCR(IROW,1)			
ATF(IROW,1)		BTF(IROW,1)		CTF(IROW,1)		DTF(IROW,1)		ATR(IROW,1)		BTR(IROW,1)		CTR(IROW,1)		DTR(IROW,1)			
ELE(IROW,1)		ETE(IROW,1)															
ACF(IROW,2)		BCF(IROW,2)		CCF(IROW,2)		DCF(IROW,2)		ACR(IROW,2)		BCR(IROW,2)		CCR(IROW,2)		DCR(IROW,2)			
ATF(IROW,2)		BTF(IROW,2)		CTF(IROW,2)		DTF(IROW,2)		ATR(IROW,2)		BTR(IROW,2)		CTR(IROW,2)		DTR(IROW,2)			
ELE(IROW,2)		ETE(IROW,2)															
ACF(IROW,NEL)		BCF(IROW,NEL)		CCF(IROW,NEL)		DCF(IROW,NEL)		ACR(IROW,NEL)		BCR(IROW,NEL)		CCR(IROW,NEL)		DCR(IROW,NEL)			
ATF(IROW,NEL)		BTF(IROW,NEL)		CTF(IROW,NEL)		DTF(IROW,NEL)		ATR(IROW,NEL)		BTR(IROW,NEL)		CTR(IROW,NEL)		DTR(IROW,NEL)			
ELE(IROW,NEL)		ETE(IROW,NEL)															

DESCRIPTION OF THE INPUT PARAMETERS

Parameter	Description	Format
AA	Type of calculating station or blade row (ANNULAR, ROTOR or STATOR). The default is ANNULAR.	A4
ACF(J)	The linear coefficient for the blade element centerline angle equation for the front segment, $\kappa = \kappa_t + ACF \times S + BCF \times S^2 + CCF \times S^3 + DCF \times S^4$ (Note: See IGEO for the definition of S)	F10.4
ACR(J)	The linear coefficient for the blade element centerline angle equation for the rear segment, $\kappa = \kappa_t + ACR \times S + BCR \times S^2 + CCR \times S^3 + DCR \times S^4$	F10.4
ADAMP	Slope angle of the damper centerline in the meridional view (deg.)	F10.4
ALP(J)	Slope angle of the blade element in the meridional view (deg.)	F10.4
ATF(J)	The first coefficient of the blade element thickness equation forward of the maximum thickness point, $\frac{t}{2c} = \frac{t_m}{2c} - ATF \left(\sqrt{S_o - S} - \sqrt{S_o} + \frac{S}{2\sqrt{S_o}} \right) - BTF \times S^2 - CTF \times S^3 - DTF \times S^4$	F10.4
ATR(J)	The first coefficient of the blade element thickness equation rearward of the maximum thickness point, $\frac{t}{2c} = \frac{t_m}{2c} - ATR \left(\sqrt{S_o - S} - \sqrt{S_o} + \frac{S}{2\sqrt{S_o}} \right) - BTR \times S^2 - CTR \times S^3 - DTR \times S^4$	F10.4
BCF(J)	The quadratic coefficient for the blade element centerline angle equation for the front segment, $\kappa = \kappa_t + ACF \times S + BCF \times S^2 + CCF \times S^3 + DCF \times S^4$	F10.4

DESCRIPTION OF THE INPUT PARAMETERS (con't)

Parameter	Description	Format
BCR(J)	The quadratic coefficient for the blade element centerline angle equation for the rear segment, $\kappa = \kappa_t + \text{ACR} \times S + \text{BCR} \times S^2 + \text{CCR} \times S^3 + \text{DCR} \times S^4$	F10.4
BH(I)	Boundary layer displacement from the hub (in.).	F10.4
BLADES	The number of blades in each rotor or stator blade row.	F10.4
BT(I)	Boundary layer displacement from the casing (in.).	F10.4
BTF(J)	Quadratic coefficient of the blade element thickness equation forward of the maximum thickness point, $\frac{t}{2c} = \frac{t_m}{2c} - \text{ATF} \left(\sqrt{S_o - S} - \sqrt{S_o} + \frac{S}{2\sqrt{S_o}} \right) - \text{BTF} \times S^2 - \text{CTF} \times S^3 - \text{DTF} \times S^4$	F10.4
BTR(J)	Quadratic coefficient of the blade element thickness equation rearward of the maximum thickness point, $\frac{t}{2c} = \frac{t_m}{2c} - \text{ATR} \left(\sqrt{S_o - S} - \sqrt{S_o} + \frac{S}{2\sqrt{S_o}} \right) - \text{BTR} \times S^2 - \text{CTR} \times S^3 - \text{DTR} \times S^4$	F10.4
CCF(J)	The cubic coefficient for the blade element centerline angle equation for the front segment, $\kappa = \kappa_t + \text{ACF} \times S + \text{BCF} \times S^2 + \text{CCF} \times S^3 + \text{DCF} \times S^4$	F10.4
CCR(J)	The cubic coefficient for the blade element centerline angle equation for the rear segment, $\kappa = \kappa_t + \text{ACR} \times S + \text{BCR} \times S^2 + \text{CCR} \times S^3 + \text{DCR} \times S^4$	F10.4
CHORD(J)	Chord length of a blade element (in.).	F10.4

DESCRIPTION OF THE INPUT PARAMETERS (con't)

Parameter	Description	Format
CDAMP	Chord length of a mid-span shroud in the meridional plane (in.).	F10.4
CPCO(I2)		
For I2 = 1,6	Constants for the specific heat polynomial function of temperature, $C_p = CPCO(1) + CPCO(2) \times T + CPCO(3) \times T^2 +$ $CPCO(4) \times T^3 + CPCO(5) \times T^4 + CPCO(6) \times T^5$	E20.8
CRENGY	An estimate of the cumulative energy addition fraction through a particular rotor to the total energy addition of the compressor. Thus, the fractions are progressively larger positive numbers through successive rotors. For the last rotor, CRENGY should be 1.0.	F10.4
CTF(J)	The cubic coefficient of the blade element thickness equation forward of the maximum thickness point, $\frac{t}{2c} = \frac{t_m}{2c} - ATF \left(\sqrt{S_o - S} - \sqrt{S_o} + \frac{S}{2\sqrt{S_o}} \right) - BTF \times S^2 - CTF \times S^3 - DTF \times S^4$	F10.4
CTR(J)	The cubic coefficient of the blade element thickness equation rearward of the maximum thickness point, $\frac{t}{2c} = \frac{t_m}{2c} - ATR \left(\sqrt{S_o - S} - \sqrt{S_o} + \frac{S}{2\sqrt{S_o}} \right) - BTR \times S^2 - CTR \times S^3 - DTR \times S^4$	F10.4
DCF(J)	The quartic coefficient for the blade element centerline angle equation for the front segment, $\kappa = \kappa_t + ACF \times S + BCF \times S^2 + CCF \times S^3 + DCF \times S^4$	F10.4
DCR(J)	The quartic coefficient for the blade element centerline angle equation for the rear segment, $\kappa = \kappa_t + ACR \times S + BCR \times S^2 + CCR \times S^3 + DCR \times S^4$	F10.4

DESCRIPTION OF THE INPUT PARAMETERS (con't)

Parameter	Description	Format
DFTAB(K,J,I)	The blade element diffusion factor (D-factor from which profile losses are tabulated. Five values are input for each streamline; that is, K always has from 1 to 5, J is the streamline index, and I is the loss set index. The maximum number of loss sets is 9. Since D-factor values normally fall between 0.3 and 0.7, values of 0.3, 0.4, 0.5, 0.6, and 0.7 for DFTAB on a streamline can be implied by leaving the DFTAB values blank. Note that you cannot set DFTAB(1,K,I) to exactly 0.0 when you do not intend to use the implied values of DFTAB.	F8.4
DLOS(K,J,I)	Profile loss parameter $\bar{w}_p \cos \frac{\theta}{2\sigma}$, associated with the DFTAB reference arrays.	F8.4
DTF(J)	The quartic coefficient of the blade element thickness equation forward of the maximum thickness point, $\frac{t}{2c} = \frac{t_m}{2c} - ATF(\sqrt{S_o - S} - \sqrt{\frac{S}{S_o}}) - BTF \times S^2 - CTF \times S^3 - DTF \times S^4$	F10.4
DTR(J)	The quartic coefficient of the blade element thickness equation rearward of the maximum thickness point, $\frac{t}{2c} = \frac{t_m}{2c} - ATR(\sqrt{S_o - S} - \sqrt{\frac{S}{S_o}}) - BTR \times S^2 - CTR \times S^3 - DTR \times S^4$	F10.4
EFF	The estimated stage adiabatic efficiency for the first iteration.	F10.4
ELE(J)	Coefficients for the leading edge ellipse ratio of semimajor to semiminor axes minus 1.0, $e = \frac{b}{a} - 1.0 = ELE$	F10.4
ETE(J)	Coefficients for the trailing edge ellipse ratio of semimajor to semiminor axes minus 1.0, $e = \frac{b}{a} - 1.0 = ETE$	F10.4

DESCRIPTION OF THE INPUT PARAMETERS (con't)

Parameter	Description	Format
FLOFRA(J)	The cumulative weight-flow split between streamlines starting from the tip. NTUBES, which is NSTRM - 1, values are read. Thus, the first value is greater than zero; and succeeding values must increase to 1.0 to make the accumulation target a given overall flow rate.	F10.4
FLOW	The initial reference flow rate (lb/sec).	F10.4
BLEED(I)	Bleed flow as fraction of flow at calculating stations; i.e., a simulation casing bleed to get a high pressure ratio compressor through the part speed area mismatch.	F8.4
I	Calculation station index. (Maximum of 100).	I3
J	Streamline index. (Maximum of 11).	I2
IDAMP	Option parameter for a part span shroud. If IDAMP > 0, a read of damper geometry information on the next line is activated.	I5
IGEO	A blade definition index. When IGEO is 0, the blade segment centerline and surfaces are defined by $d\kappa/dS = \text{Constant}$. When IGEO is not 0, the segment centerline and surfaces are defined with 4th degree functions of path distance(s) from the transition point and maximum thickness points respectively. The specification of the coefficients for these functions is extra input. If IGEO is positive, the coefficients for the definition polynomials are interpreted to be functions of segment length normalized to 1.0; but if IGEO is negative, the coefficients are interpreted to be functions of segment length normalized by chord.	I5

DESCRIPTION OF THE INPUT PARAMETERS (con't)

Parameter	Description	Format
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The reference point for the centerline polynomials can be either the transition point or the segment ends. The IGEO options are given by the following table:

Table of the IGEO Options

IGEO	Centerline		Thickness		Centerline		Thickness	
	S_1	S_2	$S_{m,1}$	$S_{m,2}$	S_1	S_2	$S_{m,1}$	$S_{m,2}$
	Origin				Range (all positive S)			
-4	Leading edge	Trailing edge	Maximum thickness	Maximum thickness	0 to $\frac{S_1}{c}$	0 to $\frac{S_2}{c}$	0 to $\frac{S_{m,1}}{c}$	0 to $\frac{S_{m,2}}{c}$
-3	Transition point	Trailing edge						
-2	Leading edge	Transition point						
-1 or < -4	Transition point	Transition point						
1 or > 4	Transition point	Transition point			0 to 1.0	0 to 1.0	0 to 1.0	0 to 1.0
2	Leading edge	Transition point						
3	Transition point	Trailing edge						
4	Leading edge	Trailing edge						

ILOSS Indicator designation of which profile loss set [I variable in DLOS(K,J,I)] to I5 use with a particular blade row.

DESCRIPTION OF THE INPUT PARAMETERS (con't)

Parameter	Description	Format
IROTOR	Rotor index.	
IROW	Blade row index.	
ITER	Iteration counter.	
ISTN(I)	Integer designation of calculating station type: 1 - Rotor inlet 2 - Rotor outlet -1 - Stator inlet -2 - Stator outlet 0 - ANNULAR	
KIC(J)	Local blade element inlet angle at the centerline (deg.).	F10.4
KOC(J)	Local blade element outlet angle at the centerline (deg.).	F10.4
MOLE	Molecular weight of the gas (28.97 for dry air).	F10.4
NA	Number of annular stations at which radial velocity profiles are constructed during computation.	I5
NEL	Number of blade definition elements in a blade row, usually 11.	I5
NHUB	Number of input points to describe the hub geometric boundary (Maximum of 100).	I5
NLOSS	Number of loss sets (Maximum of 9).	I5

DESCRIPTION OF THE INPUT PARAMETERS (con't)

Parameter	Description	Format
NBROWS	Number of blade rows (Maximum of 40).	I5
NSTRM	Number of streamlines (Maximum of 11).	I5
NSPD	Number of speed lines (Maximum of 10).	I5
NTIP	Number of input points to describe the casing geometric boundary (Maximum of 100).	I5
P(J)	Ratio of inlet segment turning rate to outlet turning rate, $[(\frac{d\psi}{ds})_1/[(\frac{d\psi}{ds})_0]]$ at the transition point of a blade element.	F10.4
PO(1,J)	General stagnation pressure array, which is in Lb/ft ² units internal to the program. I is the calculation station index; and J is the streamline index. Only [PO(1,J), J = 1, NSTRM] values are input; i.e., the streamline values for the 1st calculating station. The input values are in lb/in ²	F10.4
PR	The desired overall compressor pressure ratio.	F10.4
RANGE-CHOKE	The normalized calculation curve distance from the design point to the near choke point for each calculated speed line (a value of 0.1 to 0.2 is typical).	F10.4
RANGE-STALL	The normalized calculation curve distance from the design point to the near stall point for each calculated speed line (a value of 0.1 to 0.2 is typical).	F10.4
RDAMP	Radial location of the leading edge of a part-span shroud (in.).	F10.4
RDLE	Leading edge to chord ratio of a part-span shroud.	F10.4
RDTE	Trailing edge to chord ratio of a part-span shroud.	F10.4

DESCRIPTION OF THE INPUT PARAMETERS (con't)

Parameter	Description	Format
RESET-IGV(IP)	Guide vane reset for each speed (deg.).	
RESET-STAI(IP)	Stator 1 reset for each speed (deg.).	
RESET-STA2(IP)	Stator 2 reset for each speed (deg.).	
RESET-STA3(IP)	Stator 3 reset for each speed (deg.).	
RESET-STA4(IP)	Stator 4 reset for each speed (deg.).	
RHUB(I)	Radial coordinates (in.) of the set of points used to define the hub geometric boundary (Maximum of 100).	F10.4
ROT	The compressor rotational speed (rpm).	F10.4
RSP(J)	Radial location of blade element intersection with the blade stacking line (in.).	F10.4
RTIP(I)	Radial coordinates (in.) of the set of points used to define the casing geometric boundary (Maximum of 100).	F10.4
SPEED(IP)	Speed line values to be investigated (values are relative to the input rotational speed; i.e., 1.1, 1.0, 0.9, 0.8). (Maximum of 10).	F10.4
STEP-CHOKE	The step increment along the normalized curve distance for the choke side of each calculated speed line. The STEP value is usually about 25% of the RANGE value.	F10.4
STEP-STALL	The step increment along the normalized curve distance for the stall side of each calculated speed line. The STEP value is usually about 25% of the RANGE value.	F10.4

DESCRIPTION OF THE INPUT PARAMETERS (con't)

Parameter	Description	Format
T(J)	Transition point location between the two segments of a blade element as a fraction of chord.	F10.4
TDAMP	Maximum thickness to chord ratio of the damper.	F10.4
THLE(J)	Leading edge radius to chord ratio of a blade element.	F10.4
THMAX(J)	A blade element maximum thickness to chord ratio	F10.4
THTE(J)	Trailing edge radius to chord ratio of a blade element.	F10.4
TTILT	Angle of the stacking axis tilt (deg.) in the circumferential direction ($r - \theta$ plane). The angle is positive in the direction of rotor rotation.	F10.4
TTITLE(I)	Description of the compressor for printout.	18A4
TO(I,J)	General stagnation temperature array, which is in Deg. R units internal to the program. I is the calculation station index; and J is the streamline index. Only [TO(1,J), $J = 1$, NSTRM] values are input, i.e., the streamline values for the 1st calculating station. The input values Deg. R units.	F10.4
TS(J)	Blade element leading edge circle center angular location (θ) relative to blade stacking line (Deg.). The angle is positive in the direction of rotation.	F10.4
VTH(I,J)	General tangential component of velocity array in the program. Only the [VTH(1,J), $J = 1$, NSTRM] values are input; i.e., the streamline values for the 1st calculating station. The input values are in ft/sec.	F10.4

DESCRIPTION OF THE INPUT PARAMETERS (con't)

Parameter	Description	Format
XHUB(I)	Axial coordinates of the set of points used to define the hub geometric boundary (in.). The data range must at least reach from the furthest upstream annular calculating station to the furthest downstream calculating station. The maximum number of points is 100; and the minimum is 4.	F10.4
XTIP(I)	Axial coordinates of the set of points used to define the casing geometric boundary (in.). The conditions and limitations are the same as those on XHUB.	F10.4
ZHUB(INAB)	The hub axial coordinate (in.). When the data set is a blade rather than an ANNULAR station, ZHUB(INAB) is the axial location of the axial location of the blade stacking line at the hub.	F10.4
ZM(J)	Location of the maximum thickness point as a fraction of blade element chord.	F10.4
ZS(J)	Blade element leading edge circle center axial location relative to that of the blade element intersection with the blade stacking line (in.).	F10.4
ZTIP(INAB)	The casing axial coordinate (in.). When the data set is a blade rather than an ANNU station, ZHUB(INAB) is the axial location of the blade stacking line at the hub.	F10.4

TABLE 1 - EXAMPLE INPUT DATA SET(no stator resets)

annular	-1.00000	-1.50000	.03540	.09903	0.0000	.5000
rotor	.94100	.94100	.04571	.12239	0.0000	
11 1 0	0.000	22.0	.0524	.1026	0.0000	
.001800	.001801	.029008	.640000	.700000	.075000	
67.263710	53.819990	3.664327	-10.148210	9.855471	.742835	-10.112840
.001854	.001868	.029998	.630000	.647400	.180000	
63.414730	52.954860	3.670819	-7.508253	9.419851	.877825	-10.289330
.001984	.002026	.032414	.620000	.604200	.430000	
61.345810	52.732890	3.668617	-5.697886	8.993848	.943206	-10.565180
.002187	.002267	.036168	.610000	.562700	.660000	
59.472620	50.577800	3.666607	-3.908523	8.553914	1.004472	-10.879460
.002459	.002582	.041154	.600000	.519300	.790000	
57.361910	47.352410	3.665117	-2.152351	8.094000	1.077046	-11.211970
.002793	.002957	.047256	.580000	.470500	.830000	
55.066130	42.201420	3.664356	-3.83670	7.608888	1.158294	-11.490080
.003185	.003377	.054318	.560000	.418000	.860000	
52.454880	36.314500	3.664808	1.442794	7.090268	1.250203	-11.752500
.003623	.003818	.062085	.540000	.359200	.960000	
49.456170	27.960340	3.667629	3.459608	6.528422	1.362129	-11.896410
.004085	.004243	.070067	.500000	.286200	.980000	
45.821740	17.331430	3.675878	5.871156	5.906867	1.483685	-11.707410
.004525	.004602	.077171	.500000	.224300	1.000000	
41.865530	2.559158	3.697754	8.993676	5.197867	1.644855	-11.116670
.004797	.004796	.080222	.500000	.162900	1.000000	
36.960720	-16.395800	3.761671	13.852820	4.331938	1.802996	-9.081044
.005082	.005183	.000000	-.691483	-.696423	.000000	-.691483
.000000	.027085	.000000	.000000	.000076	.000000	.271592
.992906	.000000	.000000	-.594778	-.646003	.000000	-.594778
.348614	.594778	.000000	.000000	.000852	.000000	.256331
.000000	.029343	.000000	.000000	.000000	.000000	.000000
.920703	.000000	.000000	-.514857	-.606206	.000000	-.514857
.348777	.514857	.000000	.000000	.001702	.000000	.252549
.000000	.033250	.000000	.000000	.002757	.000000	.256919
.849311	.000000	.000000	-.437443	-.564391	.000000	-.437443
.345669	.437443	.000000	.000000	.000000	.000000	.000000
.000000	.038916	.000000	.000000	.004110	.000000	.361985
.775071	.000000	.000000	-.261985	-.519186	.000000	-.261985
.338194	.361985	.000000	.000000	.000000	.000000	.000000
.000000	.046518	.000000	.000000	.005776	.000000	.266956
.697216	.000000	.000000	-.288629	-.469716	.000000	-.288629
.325372	.288629	.000000	.000000	.000000	.000000	.000000
.000000	.058034	.000000	.000000	.000000	.000000	.000000
.614603	.000000	.000000	-.219088	-.415840	.000000	-.219088
.307296	.219088	.000000	.000000	.007863	.000000	.255666
.000000	.072581	.000000	.000000	.010427	.000000	.250194
.525593	.000000	.000000	-.153511	-.358824	.000000	-.153511
.282069	.153511	.000000	.000000	.000000	.000000	.000000
.000000	.092867	.000000	.000000	.000000	.000000	.000000
.427815	.000000	.000000	-.093499	-.294583	.000000	-.093499
.247833	.093499	.000000	.000000	.013199	.000000	.213884
.000000	.119800	.000000	.000000	.000000	.000000	.000000
.317394	.000000	.000000	.000000	.000000	.000000	.000000
.206065	.042507	.000000	-.042507	.227319	.000000	-.042507
.000000	.131741	.000000	.000000	.016899	.000000	.225236
.186993	.000000	.000000	.000000	-.002953	.000000	-.002953
.151836	.002953	.000000	.000000	-.153313	.000000	-.002953
.000000	.134921	.000000	.000000	.020384	.000000	.215452
.019264	.000000	.000000	.000000	.000000	.000000	.000000
annular	3.30000	.05534	.10998	0.0000		
stator	5.20000	.60698	.11928	0.0000		
11 2 1	0.000	34.0	.0697	.1309	0.0000	
.012897	.012878	.500000	.400000	.400000	1.000000	
36.693180	-16.091950	2.270252	-.289535	9.532545	1.053663	2.319497
.012291	.012281	.078228	.500000	.400000	1.000000	
33.456100	-12.257040	2.270269	-.237212	9.182470	1.057440	2.291867

TABLE 1 - EXAMPLE INPUT DATA SET (no stator resets)

33	.011681	.011666	.076701	.500000	.400000	1.000000	.000000	.000000	.000000
33	.488160	-10.482750	2.270352	.566314	8.825970	1.054125	2.455457	.000000	.000000
33	.011047	.011029	.075116	.500000	.400000	1.000000	.000000	.000000	.000000
33	.898910	-9.693403	2.270317	.924359	8.456358	1.051422	2.635809	.000000	.000000
34	.010385	.010367	.073463	.500000	.400000	1.000000	.000000	.000000	.000000
34	.603690	-9.097510	2.270816	1.349323	8.071085	1.047920	2.858991	.000000	.000000
35	.009687	.009675	.071119	.500000	.400000	1.000000	.000000	.000000	.000000
35	.844540	-8.800952	2.271337	1.878423	7.666330	1.042652	3.153324	.000000	.000000
37	.008944	.008943	.069861	.500000	.400000	1.000000	.000000	.000000	.000000
37	.274850	-8.604356	2.272220	2.539784	7.236899	1.036445	3.509247	.000000	.000000
38	.008146	.008162	.067864	.500000	.400000	1.000000	.000000	.000000	.000000
38	.927800	-8.807656	2.273599	3.328634	6.776628	1.029862	3.937878	.000000	.000000
40	.007277	.007317	.065693	.500000	.400000	1.000000	.000000	.000000	.000000
40	.825370	-9.008737	2.275653	4.252992	6.276914	1.021594	4.491714	.000000	.000000
43	.006312	.006374	.063281	.500000	.400000	1.000000	.000000	.000000	.000000
43	.448150	-10.287890	2.278276	5.229607	5.720603	1.012673	5.234231	.000000	.000000
46	.005201	.005230	.060502	.500000	.400000	1.000000	.000000	.000000	.000000
46	.876120	-14.152360	2.278774	5.458222	5.063943	1.008646	6.219320	.000000	.000000
334333	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.000000	.000000	.105577	.000000	.000000	.000000	.000000	.105650	.000000	.000000
.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.309119	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.000000	.000000	.106627	.000000	.000000	.000000	.000000	.106658	.000000	.000000
.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.302450	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.000000	.000000	.106211	.000000	.000000	.000000	.000000	.106252	.000000	.000000
.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.296337	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.000000	.000000	.105421	.000000	.000000	.000000	.000000	.105468	.000000	.000000
.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.290864	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.000000	.000000	.104481	.000000	.000000	.000000	.000000	.104523	.000000	.000000
.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.286973	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.000000	.000000	.103307	.000000	.000000	.000000	.000000	.103321	.000000	.000000
.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.282111	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.000000	.000000	.102012	.000000	.000000	.000000	.000000	.101972	.000000	.000000
.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.275536	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.000000	.000000	.100507	.000000	.000000	.000000	.000000	.100381	.000000	.000000
.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.267078	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.000000	.000000	.098860	.000000	.000000	.000000	.000000	.098624	.000000	.000000
.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.256524	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.000000	.000000	.096641	.000000	.000000	.000000	.000000	.096274	.000000	.000000
.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.240341	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
.000000	.000000	.093245	.000000	.000000	.000000	.000000	.092896	.000000	.000000
.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000	.000000
annular	7.34000	7.34000	.06886	.12756	0.0000	0.0000	.8526	1.0000	.0000
rotor	9.20000	9.20000	.06534	.12314	0.0000	0.0000	.8526	1.0000	.0000
11	1	0	0	0	0	0	0	0	0
.006141	.006147	.034033	.500000	.662701	.610000	.484734	-5.280378	.000000	.000000
62	.045680	56.549140	2.025434	-6.611087	9.375244	.484734	-5.280378	.000000	.000000
.006818	.006860	.035070	.500000	.635082	.663000	.501454	-5.406044	.000000	.000000
61	.064560	55.480270	2.027373	-5.160486	9.067986	.501454	-5.406044	.000000	.000000
.007507	.007589	.037438	.500000	.605315	.767000	.522647	-5.518399	.000000	.000000
59	.944920	53.732040	2.026563	-3.787091	8.754455	.522647	-5.518399	.000000	.000000
.008217	.008338	.041001	.500000	.573424	.886000	.546953	-5.630420	.000000	.000000
58	.639890	51.710970	2.025942	-2.438716	8.431924	.546953	-5.630420	.000000	.000000
.008953	.009109	.045609	.500000	.539545	.981000	.574187	-5.740963	.000000	.000000
57	.239140	49.220690	2.025542	-1.075742	8.098637	.574187	-5.740963	.000000	.000000
.008722	.008819	.041945	.500000	.501617	1.000000	.000000	.000000	.000000	.000000

TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

1

*** input data for axial-flow compressor off-design performance prediction program ***

low aspect ratio two stage nasa fan-- same as used in design report-offd

stator reset

fraction design speed	reset guide vane angle	reset stator 1 angle	reset stator 2 angle	reset stator 3 angle	reset stator 4 angle
1.00000	0.00000	0.00000	0.00000	0.00000	0.00000
0.90000	0.00000	0.00000	0.00000	0.00000	0.00000

the comp. design rotational speed is 16042.8 rpm.
the design inlet flow rate is 73.300 (lb/sec).
the compressor design pressure ratio is 2.400 .
the range to be covered to the choke side is 0.200 .
the range to be covered to the stall side is 0.050 .

calcs. will be made on 11 streamlines.
the compressor has 4 blade rows.
the molecular weight is 28.96 .
the choke point interval is 0.029 .
the stall point interval is 0.020 .

calculations will be made at the blade edges and at 17 annular stations.

the fractions of design speed to be investigated are 1.000 0.900

the specific heat polynomial is in the following form

$$cp = 0.23762E+00 + 0.39557E-04*t + -0.28463E-06*t**2 + 0.81651E-09*t**3 + -0.81994E-12*t**4 + 0.28443E-15*t**5$$

input distributions by streamline or streamtube

streamline no.	inlet total temperature (deg. F.)	inlet total pressure (psia)	inlet whirl velocity (ft/sec)	streamtube no.	streamtube flow fraction
1	518.700	14.125	0.000	1	0.1000
2	518.700	14.670	0.000	2	0.2000
3	518.700	14.700	0.000	3	0.3000
4	518.700	14.700	0.000	4	0.4000
5	518.700	14.700	0.000	5	0.5000
6	518.700	14.700	0.000	6	0.6000
7	518.700	14.700	0.000	7	0.7000
8	518.700	14.700	0.000	8	0.8000
9	518.700	14.700	0.000	9	0.9000
10	518.700	14.700	0.000	10	1.0000

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OF POOR QUALITY

TABLE 2 - EXAMPLE OUTPUT DATA SET (no statior reset)

pct. pass.	d-factor	loss param.	d-factor	loss param.	d-factor	loss param.	d-factor	loss param.	d-factor	loss param.
0.00	0.3000	0.0309	0.4000	0.0336	0.5000	0.0373	0.6000	0.0430	0.7000	0.0508
10.00	0.3000	0.0272	0.4000	0.0290	0.5000	0.0320	0.6000	0.0362	0.7000	0.0423
20.00	0.3000	0.0250	0.4000	0.0263	0.5000	0.0282	0.6000	0.0313	0.7000	0.0360
30.00	0.3000	0.0230	0.4000	0.0239	0.5000	0.0253	0.6000	0.0280	0.7000	0.0310
40.00	0.3000	0.0211	0.4000	0.0220	0.5000	0.0234	0.6000	0.0261	0.7000	0.0256
50.00	0.3000	0.0212	0.4000	0.0222	0.5000	0.0236	0.6000	0.0264	0.7000	0.0239
60.00	0.3000	0.0214	0.4000	0.0226	0.5000	0.0241	0.6000	0.0269	0.7000	0.0206
70.00	0.3000	0.0218	0.4000	0.0231	0.5000	0.0248	0.6000	0.0278	0.7000	0.0317
80.00	0.3000	0.0233	0.4000	0.0248	0.5000	0.0270	0.6000	0.0303	0.7000	0.0347
90.00	0.3000	0.0272	0.4000	0.0290	0.5000	0.0320	0.6000	0.0362	0.7000	0.0423
100.00	0.3000	0.0294	0.4000	0.0317	0.5000	0.0358	0.6000	0.0441	0.7000	0.0486

1

*** input station data ***

** input set no. 1 is an annular station **

tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)
-11.0000	-11.0000	0.0000	0.0000

** input set no. 2 is an annular station **

tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)
-9.0000	-9.0000	0.0044	0.0122

** input set no. 3 is an annular station **

tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)
-7.0000	-7.0000	0.0088	0.0249

** input set no. 4 is an annular station **

tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)
-5.2000	-5.2000	0.0133	0.0382

** input set no. 5 is an annular station **

tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)
-3.7000	-3.7000	0.0222	0.0632

** input set no. 6 is an annular station **

tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)

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TABLE 2 - EXAMPLE OUTPUT DATA SET (no statior reset)

1	0.01290	0.01288	0.07974	0.50000	0.40000	1.00000	35.593	-15.092	2.27025	-0.290	9.5325	1.05356	2.319
2	0.01229	0.01228	0.07823	0.50000	0.40000	1.00000	33.496	-12.257	2.27027	0.237	9.1825	1.05744	2.292
3	0.01168	0.01167	0.07670	0.50000	0.40000	1.00000	33.488	-10.483	2.27035	0.567	8.8260	1.05412	2.435
4	0.01105	0.01103	0.07512	0.50000	0.40000	1.00000	33.899	-9.093	2.27052	0.925	8.4564	1.05142	2.636
5	0.01039	0.01037	0.07346	0.50000	0.40000	1.00000	34.604	-8.098	2.27082	1.349	8.0711	1.04792	2.859
6	0.00969	0.00967	0.07172	0.50000	0.40000	1.00000	35.845	-8.801	2.27134	1.878	7.6663	1.04265	3.153
7	0.00894	0.00894	0.06986	0.50000	0.40000	1.00000	37.275	-8.808	2.27222	2.540	7.2369	1.03645	3.509
8	0.00815	0.00816	0.06786	0.50000	0.40000	1.00000	38.928	-9.009	2.27360	3.329	6.7766	1.02986	3.938
9	0.00728	0.00732	0.06569	0.50000	0.40000	1.00000	40.825	-9.299	2.27565	4.253	6.2769	1.02159	4.492
10	0.00631	0.00637	0.06328	0.50000	0.40000	1.00000	43.448	-10.288	2.27828	5.230	5.7206	1.01267	5.234
11	0.00520	0.00523	0.06050	0.50000	0.40000	1.00000	46.876	-14.132	2.27877	5.458	5.0639	1.00865	6.219
* parameters for greater specification of blade element geometry *													
poly. coef. for 1st seg. centerline angle													
(function of path dist. from trans. pt.)													

linear quadratic cubic quartic													
1	0.33433	0.00000	0.00000	0.00000	0.00000	0.00000	-0.33433	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
2	0.30912	0.00000	0.00000	0.00000	0.00000	0.00000	-0.30912	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
3	0.30245	0.00000	0.00000	0.00000	0.00000	0.00000	-0.30245	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
4	0.29634	0.00000	0.00000	0.00000	0.00000	0.00000	-0.29634	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
5	0.29086	0.00000	0.00000	0.00000	0.00000	0.00000	-0.29086	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
6	0.28697	0.00000	0.00000	0.00000	0.00000	0.00000	-0.28697	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
7	0.28211	0.00000	0.00000	0.00000	0.00000	0.00000	-0.28211	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
8	0.27554	0.00000	0.00000	0.00000	0.00000	0.00000	-0.27554	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
9	0.26708	0.00000	0.00000	0.00000	0.00000	0.00000	-0.26708	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
10	0.25652	0.00000	0.00000	0.00000	0.00000	0.00000	-0.25652	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
11	0.24034	0.00000	0.00000	0.00000	0.00000	0.00000	-0.24034	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
poly. coef. for 2nd seg. centerline angle													
(function of path dist. from trans. pt.)													

linear quadratic cubic quartic													
1	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000	0.00000
2	0.00000	0.10558	0.00000	0.00000	0.00000	0.00000	0.00000	0.10555	0.00000	0.00000	0.00000	0.00000	0.00000
3	0.00000	0.10663	0.00000	0.00000	0.00000	0.00000	0.00000	0.10666	0.00000	0.00000	0.00000	0.00000	0.00000
4	0.00000	0.10521	0.00000	0.00000	0.00000	0.00000	0.00000	0.10625	0.00000	0.00000	0.00000	0.00000	0.00000
5	0.00000	0.10542	0.00000	0.00000	0.00000	0.00000	0.00000	0.10547	0.00000	0.00000	0.00000	0.00000	0.00000
6	0.00000	0.10446	0.00000	0.00000	0.00000	0.00000	0.00000	0.10452	0.00000	0.00000	0.00000	0.00000	0.00000
7	0.00000	0.10331	0.00000	0.00000	0.00000	0.00000	0.00000	0.10332	0.00000	0.00000	0.00000	0.00000	0.00000
8	0.00000	0.10291	0.00000	0.00000	0.00000	0.00000	0.00000	0.10197	0.00000	0.00000	0.00000	0.00000	0.00000
9	0.00000	0.10051	0.00000	0.00000	0.00000	0.00000	0.00000	0.10038	0.00000	0.00000	0.00000	0.00000	0.00000
10	0.00000	0.09886	0.00000	0.00000	0.00000	0.00000	0.00000	0.09862	0.00000	0.00000	0.00000	0.00000	0.00000
11	0.00000	0.09664	0.00000	0.00000	0.00000	0.00000	0.00000	0.09627	0.00000	0.00000	0.00000	0.00000	0.00000
*** input station data ***													
** input set no. 11 is an annular station **													
strm. no.	tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)									
1	7.3400	7.3400	0.0689	0.1276									
*** input station data ***													
** input set no. 12 is rotor no. 2 **													

TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

tip c.g. axial location (inches)	hub c.g. axial location (inches)	inlet tip b. layer thick. (inches)	inlet hub b. layer thick. (inches)	blade dihedral angle (degrees)	range improvement bleed option
9.2000	9.2000	0.0663	0.1231	0.00	0.00000

number of blades	media. eff. of cum. comp. stages at design	outlet tip b. layer thick. (inches)	outlet hub b. layer thick. (inches)	cum. energy addition fract. thru this stage
38	0.8526	0.0608	0.0995	1.0000

* table of input blade element definition parameters *

strm. no.	l.e.rad. /chord	t.e.rad. /chord	max.th. /chord	max.th. pt.loc.	tran.pt. /chord	segment in/out	in-blade turn.rate	angle (deg)	aero. chord (in.)	layout cone ang. (deg)	stack pt. radius (in.)	l.e.cir.cen.loc. from stack. line z (in.) o (deg)
1	0.00614	0.00615	0.03403	0.50000	0.66270	0.61000	62.046	56.549	2.02543	-6.611	9.3752	0.48473
2	0.00682	0.00686	0.03507	0.50000	0.63508	0.63300	61.065	55.480	2.02737	-5.160	9.0680	0.50145
3	0.00751	0.00759	0.03744	0.50000	0.60532	0.67000	59.945	53.732	2.02656	-3.787	8.7545	0.52265
4	0.00822	0.00834	0.04100	0.50000	0.57342	0.88600	58.660	51.711	2.02594	-2.439	8.4319	0.54695
5	0.00895	0.00911	0.04561	0.50000	0.53954	0.98100	57.239	49.231	2.02554	-1.076	8.0986	0.57419
6	0.00972	0.00991	0.05110	0.50000	0.50394	1.00000	55.718	46.031	2.02544	0.303	7.7514	0.60400
7	0.01053	0.01075	0.05726	0.50000	0.46654	1.00000	54.155	41.968	2.02578	1.746	7.3862	0.63685
8	0.01140	0.01163	0.06383	0.50000	0.42725	1.00000	52.589	36.760	2.02689	3.363	6.9982	0.67351
9	0.01235	0.01256	0.07038	0.50000	0.40000	1.00000	51.128	30.009	2.02957	5.318	6.5811	0.71444
10	0.01341	0.01357	0.07622	0.50000	0.40000	1.00000	50.352	20.164	2.03577	7.811	6.1192	0.76065
11	0.01474	0.01476	0.07989	0.50000	0.40000	1.00000	52.662	-0.228	2.05176	11.213	5.5584	0.83368

*** input station data ***

** input set no. 13 is an annular station **

tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)
11.0100	11.0100	0.0577	0.0928

*** input station data ***

** input set no. 14 is a guide vane or stator **

tip c.g. axial location (inches)	hub c.g. axial location (inches)	inlet tip b. layer thick. (inches)	inlet hub b. layer thick. (inches)	blade dihedral angle (degrees)	range improvement bleed option
12.7000	12.7000	0.0564	0.0890	0.00	0.00000

number of blades	blade rotation option for stall margin	outlet tip b. layer thick. (inches)	outlet hub b. layer thick. (inches)
42	0.0556	0.0866	0.0866

TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

* table of input blade element definition parameters *														
strm. no.	l.e. rad. /chord	t.e. rad. /chord	max th. /chord	max th. pt. loc. /chord	tran.pt. location /chord	segment in/out	in-blade angle (deg)	out-blade angle (deg)	aero. chord (in.)	layout cone ang. (deg)	stack pt. radius (in.)	l.e. cir. cen. loc. from stack. line z (in.)	o (deg)	
1	0.01387	0.01387	0.07967	0.50000	0.40000	1.00000	35.156	-15.591	1.75296	0.218	9.2437	0.81577	1.758	
2	0.01323	0.01323	0.07807	0.50000	0.40000	1.00000	34.571	-12.796	1.75301	0.413	8.9689	0.81297	1.857	
3	0.01257	0.01257	0.07643	0.50000	0.40000	1.00000	34.651	-10.900	1.75305	0.584	8.6891	0.80990	1.981	
4	0.01190	0.01190	0.07476	0.50000	0.40000	1.00000	34.762	-9.901	1.75310	0.743	8.4021	0.80850	2.091	
5	0.01121	0.01121	0.07304	0.50000	0.40000	1.00000	35.247	-9.401	1.75316	0.900	8.1069	0.80689	2.222	
6	0.01050	0.01049	0.07125	0.50000	0.40000	1.00000	36.187	-9.202	1.75324	1.077	7.8010	0.80427	2.390	
7	0.00975	0.00975	0.06938	0.50000	0.40000	1.00000	37.386	-9.103	1.75335	1.280	7.4815	0.80084	2.594	
8	0.00897	0.00896	0.06742	0.50000	0.40000	1.00000	38.803	-9.304	1.75349	1.497	7.1458	0.79710	2.831	
9	0.00814	0.00813	0.06535	0.50000	0.40000	1.00000	40.462	-9.603	1.75364	1.703	6.7902	0.79249	3.117	
10	0.00724	0.00723	0.06311	0.50000	0.40000	1.00000	43.769	-11.099	1.75379	1.906	6.4061	0.78390	3.556	
11	0.00621	0.00620	0.06051	0.50000	0.40000	1.00000	52.207	-15.990	1.75422	2.433	5.9636	0.76049	4.456	
*** input station data ***														
** input set no. 15 is an annular station **														
			tip axial location (inches)	14.4400				hub axial location (inches)	14.4400				tip b. layer thick. (inches)	0.0552
			hub b. layer thick. (inches)											
** input set no. 16 is an annular station **														
			tip axial location (inches)	15.7000				hub axial location (inches)	16.0000				tip b. layer thick. (inches)	0.0549
			hub b. layer thick. (inches)											
** input set no. 17 is an annular station **														
			tip axial location (inches)	17.0000				hub axial location (inches)	17.6000				tip b. layer thick. (inches)	0.0538
			hub b. layer thick. (inches)											
** input set no. 18 is an annular station **														
			tip axial location (inches)	17.7500				hub axial location (inches)	18.6000				tip b. layer thick. (inches)	0.0529
			hub b. layer thick. (inches)											
** input set no. 19 is an annular station **														
			tip axial location (inches)					hub axial location (inches)					tip b. layer thick. (inches)	
			hub b. layer thick. (inches)											

TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

18.5000		19.6000		0.0521		0.0730					
** input set no. 20 is an annular station **											
tip axial location (inches)		hub axial location (inches)		tip b. layer thick. (inches)		hub b. layer thick. (inches)					
19.2500		20.6000		0.0511		0.0687					
*** input station data ***											
** input set no. 21 is an annular station **											
tip axial location (inches)		hub axial location (inches)		tip b. layer thick. (inches)		hub b. layer thick. (inches)					
20.0000		21.5000		0.0500		0.0649					
radius array											
i	1	2	3	4	5	6	7	8	9	10	11
1	10.0990	9.6525	9.1843	8.6909	8.1678	7.6088	7.0053	6.3447	5.6067	4.7556	3.7144
2	10.0948	9.6462	9.1758	8.6799	8.1539	7.5915	6.9840	6.3184	5.5738	4.7130	3.6548
3	10.0912	9.6407	9.1680	8.6697	8.1409	7.5753	6.9639	6.2933	5.5422	4.6720	3.5969
4	10.0878	9.6356	9.1611	8.6606	8.1294	7.5609	6.9461	6.2713	5.5145	4.6358	3.5457
5	10.0787	9.6283	9.1558	8.6575	8.1288	7.5632	6.9517	6.2809	5.5294	4.6582	3.5810
6	10.0698	9.6215	9.1513	8.6556	8.1297	7.5673	6.9596	6.2935	5.5480	4.6854	3.6229
7	10.0652	9.6219	9.1573	8.6677	8.1488	7.5945	6.9964	6.3422	5.6123	4.7719	3.7477
8	10.0542	9.6149	9.1545	8.6696	8.1560	7.6078	7.0170	6.3715	5.6528	4.8283	3.8302
9	9.5746	9.3116	8.9202	8.5109	8.0808	7.6265	7.1434	6.6251	6.0627	5.4425	4.7419
10	9.5746	9.3116	8.9202	8.5109	8.0808	7.6265	7.1434	6.6251	6.0627	5.4425	4.7419
11	9.5376	9.1835	8.8151	8.4307	8.0280	7.6039	7.1547	6.6753	6.1588	5.5947	4.9671
12	9.5268	9.1849	8.8297	8.4597	8.0727	7.6662	7.2370	6.7806	6.2911	5.7603	5.1752
13	9.5337	9.1961	8.8456	8.4807	8.0993	7.6991	7.2769	6.8287	6.3489	5.8297	5.2595
14	9.4327	9.1096	8.7747	8.4264	8.0631	7.6826	7.2823	6.8587	6.4071	5.9212	5.3916
15	9.3129	9.0199	8.7171	8.4033	8.0774	7.7378	7.3825	7.0093	6.6150	6.1957	5.7459
16	9.2570	8.9751	8.6841	8.3830	8.0706	7.7457	7.4065	7.0511	6.6767	6.2800	5.8566
17	9.2405	8.9647	8.6801	8.3859	8.0810	7.7641	7.4338	7.0880	6.7245	6.3402	5.9310
18	9.2471	8.9785	8.6954	8.4060	8.1063	7.7951	7.4709	7.1320	6.7761	6.4005	6.0015
19	9.2481	8.9785	8.7006	8.4135	8.1162	7.8023	7.4864	7.1507	6.7985	6.4269	6.0326
20	9.2428	8.9771	8.7032	8.4204	8.1278	7.8247	7.5085	7.1788	6.8333	6.4693	6.0836
21	9.2657	9.0097	8.7461	8.4744	8.1937	7.9030	7.6012	7.2869	6.9584	6.6136	6.2499
22	9.3553	9.1087	8.8552	8.5942	8.3251	8.0470	7.7589	7.4597	7.1479	6.8220	6.4797
23	9.5226	9.2873	9.0459	8.7979	8.5426	8.2795	8.0078	7.7265	7.4346	7.1307	6.8133
24	9.7661	9.5467	9.3221	9.0919	8.8558	8.6132	8.3635	8.1062	7.8404	7.5653	7.2798
25	10.1062	9.9008	9.6911	9.4767	9.2573	9.0326	8.8022	8.5656	8.3223	8.0716	7.8129

i	1	2	3	4	5	6
1	1	1	1	1	1	1
2	1	1	1	1	1	1
3	1	1	1	1	1	1
4	1	1	1	1	1	1
5	1	1	1	1	1	1
6	1	1	1	1	1	1

TABLE 2 - EXAMPLE OUTPUT DATA SET (no statior reset)

z array																								
i	*	1	2	3	4	5	6	7	8	9	10	11												
1	*	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000
2	*	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000
3	*	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000
4	*	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000
5	*	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000
6	*	-2.3013	-2.3218	-2.3433	-2.3650	-2.3901	-2.4158	-2.4436	-2.4741	-2.5081	-2.5476	-2.5962	-2.6522	-2.7149	-2.7848	-2.8623	-2.9478	-3.0413	-3.1428	-3.2522	-3.3693	-3.4938	-3.6257	-3.7648
7	*	-1.0027	-1.0731	-1.1512	-1.2360	-1.3274	-1.4245	-1.5273	-1.6357	-1.7494	-1.8683	-1.9922	-2.1210	-2.2546	-2.3928	-2.5355	-2.6826	-2.8340	-2.9896	-3.1493	-3.3130	-3.4806	-3.6522	-3.8277
8	*	-0.9429	-0.7890	-0.6785	-0.6087	-0.5633	-0.5337	-0.5090	-0.4893	-0.4731	-0.4598	-0.4488	-0.4396	-0.4320	-0.4258	-0.4208	-0.4268	-0.4336	-0.4409	-0.4485	-0.4563	-0.4642	-0.4722	-0.4802
9	*	1.5889	2.0072	2.2943	2.4444	2.5573	2.6324	2.7209	2.7631	2.7953	2.8169	2.8286	2.8302	2.8318	2.8334	2.8350	2.8366	2.8382	2.8398	2.8414	2.8430	2.8446	2.8462	2.8478
10	*	3.0034	3.0257	3.0489	3.0731	3.0985	3.1253	3.1537	3.1841	3.2169	3.2523	3.2902	3.3305	3.3732	3.4182	3.4654	3.5147	3.5660	3.6192	3.6742	3.7309	3.7892	3.8481	3.9075
11	*	4.1826	4.1922	4.1947	4.1959	4.1964	4.1961	4.1955	4.1945	4.1935	4.1925	4.1915	4.1905	4.1895	4.1885	4.1875	4.1865	4.1855	4.1845	4.1835	4.1825	4.1815	4.1805	4.1795
12	*	6.3435	6.3326	6.3226	6.3131	6.3046	6.2967	6.2893	6.2824	6.2760	6.2699	6.2641	6.2585	6.2531	6.2478	6.2426	6.2375	6.2324	6.2273	6.2222	6.2171	6.2120	6.2069	6.2018
13	*	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400
14	*	8.7029	8.5845	8.4610	8.3340	8.2037	8.0705	7.9341	7.7946	7.6519	7.5059	7.3564	7.2033	7.0464	6.8856	6.7208	6.5520	6.3791	6.2020	6.0206	5.8348	5.6445	5.4496	5.2500
15	*	9.7328	9.7515	9.7763	9.8038	9.8351	9.8699	9.9079	9.9481	9.9904	10.0347	10.0809	10.1288	10.1782	10.2290	10.2811	10.3344	10.3888	10.4442	10.4995	10.5557	10.6127	10.6704	10.7287
16	*	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100
17	*	11.8599	11.8638	11.8681	11.8727	11.8776	11.8824	11.8872	11.8920	11.8967	11.9013	11.9058	11.9102	11.9145	11.9187	11.9228	11.9268	11.9307	11.9345	11.9382	11.9418	11.9453	11.9487	11.9520
18	*	13.5877	13.5857	13.5840	13.5820	13.5825	13.5824	13.5824	13.5823	13.5822	13.5821	13.5820	13.5819	13.5818	13.5817	13.5816	13.5815	13.5814	13.5813	13.5812	13.5811	13.5810	13.5809	13.5808
19	*	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400
20	*	15.7050	15.7292	15.7541	15.7798	15.8064	15.8340	15.8628	15.8928	15.9242	15.9573	15.9924	16.0292	16.0667	16.1048	16.1434	16.1825	16.2220	16.2619	16.3022	16.3428	16.3837	16.4248	16.4660
21	*	17.0102	17.0590	17.1092	17.1600	17.2114	17.2639	17.3174	17.3718	17.4270	17.4828	17.5391	17.5958	17.6528	17.7100	17.7673	17.8247	17.8821	17.9394	17.9966	18.0537	18.1107	18.1675	18.2241
22	*	17.7645	17.8344	17.9062	17.9801	18.0563	18.1351	18.2167	18.3015	18.3898	18.4822	18.5791	18.6804	18.7859	18.8964	19.0118	19.1320	19.2570	19.3867	19.5210	19.6598	19.8030	19.9505	20.1024
23	*	18.5190	18.6106	18.7046	18.8011	18.9004	19.0028	19.1086	19.2181	19.3317	19.4500	19.5735	19.7014	19.8336	19.9700	20.1104	20.2547	20.4028	20.5547	20.7104	20.8697	21.0325	21.1987	21.3682
24	*	19.2737	19.3879	19.5049	19.6247	19.7477	19.8740	20.0040	20.1380	20.2764	20.4196	20.5683	20.7214	20.8788	21.0404	21.2060	21.3756	21.5491	21.7264	21.9075	22.0924	22.2810	22.4732	22.6689
25	*	20.0268	20.1357	20.2473	20.3618	20.4791	20.5991	20.7214	20.8449	20.9733	21.1059	21.2424	21.3827	21.5267	21.6743	21.8254	21.9799	22.1378	22.2990	22.4634	22.6310	22.7999	22.9710	23.1452

to array

TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

[illegible]

TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80	81	82	83	84	85	86	87	88	89	90	91	92	93	94	95	96	97	98	99	100	101	102	103	104	105	106	107	108	109	110	111	112	113	114	115	116	117	118	119	120	121	122	123	124	125	126	127	128	129	130	131	132	133	134	135	136	137	138	139	140	141	142	143	144	145	146	147	148	149	150	151	152	153	154	155	156	157	158	159	160	161	162	163	164	165	166	167	168	169	170	171	172	173	174	175	176	177	178	179	180	181	182	183	184	185	186	187	188	189	190	191	192	193	194	195	196	197	198	199	200	201	202	203	204	205	206	207	208	209	210	211	212	213	214	215	216	217	218	219	220	221	222	223	224	225	226	227	228	229	230	231	232	233	234	235	236	237	238	239	240	241	242	243	244	245	246	247	248	249	250	251	252	253	254	255	256	257	258	259	260	261	262	263	264	265	266	267	268	269	270	271	272	273	274	275	276	277	278	279	280	281	282	283	284	285	286	287	288	289	290	291	292	293	294	295	296	297	298	299	300	301	302	303	304	305	306	307	308	309	310	311	312	313	314	315	316	317	318	319	320	321	322	323	324	325	326	327	328	329	330	331	332	333	334	335	336	337	338	339	340	341	342	343	344	345	346	347	348	349	350	351	352	353	354	355	356	357	358	359	360	361	362	363	364	365	366	367	368	369	370	371	372	373	374	375	376	377	378	379	380	381	382	383	384	385	386	387	388	389	390	391	392	393	394	395	396	397	398	399	400	401	402	403	404	405	406	407	408	409	410	411	412	413	414	415	416	417	418	419	420	421	422	423	424	425	426	427	428	429	430	431	432	433	434	435	436	437	438	439	440	441	442	443	444	445	446	447	448	449	450	451	452	453	454	455	456	457	458	459	460	461	462	463	464	465	466	467	468	469	470	471	472	473	474	475	476	477	478	479	480	481	482	483	484	485	486	487	488	489	490	491	492	493	494	495	496	497	498	499	500	501	502	503	504	505	506	507	508	509	510	511	512	513	514	515	516	517	518	519	520	521	522	523	524	525	526	527	528	529	530	531	532	533	534	535	536	537	538	539	540	541	542	543	544	545	546	547	548	549	550	551	552	553	554	555	556	557	558	559	560	561	562	563	564	565	566	567	568	569	570	571	572	573	574	575	576	577	578	579	580	581	582	583	584	585	586	587	588	589	590	591	592	593	594	595	596	597	598	599	600	601	602	603	604	605	606	607	608	609	610	611	612	613	614	615	616	617	618	619	620	621	622	623	624	625	626	627	628	629	630	631	632	633	634	635	636	637	638	639	640	641	642	643	644	645	646	647	648	649	650	651	652	653	654	655	656	657	658	659	660	661	662	663	664	665	666	667	668	669	670	671	672	673	674	675	676	677	678	679	680	681	682	683	684	685	686	687	688	689	690	691	692	693	694	695	696	697	698	699	700	701	702	703	704	705	706	707	708	709	710	711	712	713	714	715	716	717	718	719	720	721	722	723	724	725	726	727	728	729	730	731	732	733	734	735	736	737	738	739	740	741	742	743	744	745	746	747	748	749	750	751	752	753	754	755	756	757	758	759	760	761	762	763	764	765	766	767	768	769	770	771	772	773	774	775	776	777	778	779	780	781	782	783	784	785	786	787	788	789	790	791	792	793	794	795	796	797	798	799	800	801	802	803	804	805	806	807	808	809	810	811	812	813	814	815	816	817	818	819	820	821	822	823	824	825	826	827	828	829	830	831	832	833	834	835	836	837	838	839	840	841	842	843	844	845	846	847	848	849	850	851	852	853	854	855	856	857	858	859	860	861	862	863	864	865	866	867	868	869	870	871	872	873	874	875	876	877	878	879	880	881	882	883	884	885	886	887	888	889	890	891	892	893	894	895	896	897	898	899	900	901	902	903	904	905	906	907	908	909	910	911	912	913	914	915	916	917	918	919	920	921	922	923	924	925	926	927	928	929	930	931	932	933	934	935	936	937	938	939	940	941	942	943	944	945	946	947	948	949	950	951	952	953	954	955	956	957	958	959	960	961	962	963	964	965	966	967	968	969	970	971	972	973	974	975	976	977	978	979	980	981	982	983	984	985	986	987	988	989	990	991	992	993	994	995	996	997	998	999	1000	1001	1002	1003	1004	1005	1006	1007	1008	1009	1010	1011	1012	1013	1014	1015	1016	1017	1018	1019	1020	1021	1022	1023	1024	1025	1026	1027	1028	1029	1030	1031	1032	1033	1034	1035	1036	1037	1038	1039	1040	1041	1042	1043	1044	1045	1046	1047	1048	1049	1050	1051	1052	1053	1054	1055	1056	1057	1058	1059	1060	1061	1062	1063	1064	1065	1066	1067	1068	1069	1070	1071	1072	1073	1074	1075	1076	1077	1078	1079	1080	1081	1082	1083	1084	1085	1086	1087	1088	1089	1090	1091	1092	1093	1094	1095	1096	1097	1098	1099	1100	1101	1102	1103	1104	1105	1106	1107	1108	1109	1110	1111	1112	1113	1114	1115	1116	1117	1118	1119	1120	1121	1122	1123	1124	1125	1126	1127	1128	1129	1130	1131	1132	1133	1134	1135	1136	1137	1138	1139	1140	1141	1142	1143	1144	1145	1146	1147	1148	1149	1150	1151	1152	1153	1154	1155	1156	1157	1158	1159	1160	1161	1162	1163	1164	1165	1166	1167	1168	1169	1170	1171	1172	1173	1174	1175	1176	1177	1178	1179	1180	1181	1182	1183	1184	1185	1186	1187	1188	1189	1190	1191	1192	1193	1194	1195	1196	1197	1198	1199	1200	1201	1202	1203	1204	1205	1206	1207	1208	1209	1210	1211	1212	1213	1214	1215	1216	1217	1218	1219	1220	1221	1222	1223	1224	1225	1226	1227	1228	1229	1230	1231	1232	1233	1234	1235	1236	1237	1238	1239	1240	1241	1242	1243	1244	1245	1246	1247	1248	1249	1250	1251	1252	1253	1254	1255	1256	1257	1258	1259	1260	1261	1262	1263	1264	1265	1266	1267	1268	1269	1270	1271	1272	1273	1274	1275	1276	1277	1278	1279	1280	1281	1282	1283	1284	1285	1286	1287	1288	1289	1290	1291	1292	1293	1294	1295	1296	1297	1298	1299	1300	1301	1302	1303	1304	1305	1306	1307	1308	1309	1310	1311	1312	1313	1314	1315	1316	1317	1318	1319	1320	1321	1322	1323	1324	1325	1326	1327	1328	1329	1330	1331	1332	1333	1334	1335	1336	1337	1338	1339	1340	1341	1342	1343	1344	1345	1346	1347	1348	1349	1350	1351	1352	1353	1354	1355	1356	1357	1358	1359	1360	1361	1362	1363	1364	1365	1366	1367	1368	1369	1370	1371	1372	1373	1374	1375	1376	1377	1378	1379	1380	1381	1382	1383	1384	1385	1386	1387	1388	1389	1390	1391	1392	1393	1394	1395	1396	1397	1398	1399	1400	1401	1402	1403	1404	1405	1406	1407	1408	1409	1410	1411	1412	1413	1414	1415	1416	1417	1418	1419	1420	1421	1422	1423	1424	1425	1426	1427	1428	1429	1430	1431	1432	1433	1434	1435	1436	1437	1438	1439	1440	1441	1442	1443	1444	1445	1446	1447	1448	1449	1450	1451	1452	1453	1454	1455	1456	1457	1458	1459	1460	1461	1462	1463	1464	1465	1466	1467	1468	1469	1470	1471	1472	1473	1474	1475	1476	1477	1478	1479	1480	1481	1482	1483	1484	1485	1486	1487	1488	1489	1490	1491	1492	1493	1494	1495
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TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25
10 *	-0.0437	-0.0594	-0.0623	-0.0536	-0.0393	-0.0197	0.0053	0.0357	0.0667	0.1052	0.1863					
11 *	-0.0169	-0.0234	-0.0245	-0.0196	-0.0113	0.0004	0.0161	0.0357	0.0569	0.0833	0.1271					
12 *	-0.0143	-0.0217	-0.0286	-0.0347	-0.0407	0.0465	0.0523	0.0581	0.0647	0.0721	0.0786					
13 *	-0.0202	-0.0184	-0.0156	-0.0113	-0.0055	0.0023	0.0125	0.0259	0.0433	0.0668	0.1013					
14 *	-0.1177	-0.0911	-0.0646	-0.0385	-0.0129	0.0127	0.0387	0.0658	0.0954	0.1293	0.1711					
15 *	-0.0882	-0.0639	-0.0396	-0.0160	0.0071	0.0298	0.0525	0.0756	0.1005	0.1293	0.1632					
16 *	-0.0228	-0.0183	-0.0132	-0.0071	0.0003	0.0092	0.0200	0.0334	0.0502	0.0718	0.1005					
17 *	-0.0109	-0.0042	-0.0028	-0.0009	0.0037	0.0246	0.0323	0.0402	0.0481	0.0558	0.0620					
18 *	-0.0054	0.0071	0.0089	0.0109	0.0131	0.0156	0.0184	0.0218	0.0257	0.0304	0.0368					
19 *	-0.0017	0.0008	0.0035	0.0062	0.0091	0.0122	0.0155	0.0191	0.0230	0.0273	0.0323					
20 *	-0.0069	-0.0022	0.0026	0.0075	0.0126	0.0180	0.0236	0.0295	0.0358	0.0427	0.0503					
21 *	0.0697	0.0783	0.0872	0.0963	0.1058	0.1157	0.1262	0.1374	0.1495	0.1627	0.1774					
22 *	0.1708	0.1792	0.1879	0.1969	0.2064	0.2165	0.2272	0.2388	0.2514	0.2654	0.2812					
23 *	0.2887	0.2783	0.2683	0.2587	0.2497	0.2416	0.2342	0.2272	0.2218	0.2174	0.2142					
24 *	0.3879	0.3991	0.4108	0.4231	0.4362	0.4501	0.4650	0.4812	0.4990	0.5185	0.5403					
25 *	0.5024	0.5110	0.5203	0.5304	0.5415	0.5537	0.5671	0.5819	0.5983	0.6166	0.6373					

rcurve array

i *	1	2	3	4	5	6	7	8	9	10	11
1 *	0.0001	0.0001	0.0002	0.0002	0.0003	0.0004	0.0005	0.0006	0.0007	0.0009	0.0010
2 *	0.0002	0.0001	0.0001	0.0001	0.0001	0.0000	0.0000	0.0000	-0.0001	-0.0002	0.0010
3 *	0.0001	0.0002	0.0002	0.0006	0.0010	0.0016	0.0022	0.0030	0.0040	0.0055	-0.0066
4 *	-0.0021	-0.0013	-0.0003	0.0007	0.0018	0.0031	0.0046	0.0065	0.0090	0.0127	0.0432
5 *	-0.0049	0.0022	0.0026	0.0034	0.0044	0.0058	0.0075	0.0089	0.0132	0.0184	-0.0031
6 *	0.0098	-0.0045	-0.0045	-0.0002	0.0039	0.0082	0.0130	0.0187	0.0263	0.0368	0.0762
7 *	-0.1086	-0.0227	-0.0158	-0.0106	-0.0057	-0.0006	0.0048	0.0140	0.0184	0.0218	0.0614
8 *	-0.1046	-0.0400	-0.0277	-0.0200	-0.0136	-0.0077	-0.0018	0.0045	0.0124	0.0218	0.0432
9 *	0.0586	0.0047	0.0037	0.0026	0.0014	0.0001	-0.0014	-0.0033	-0.0060	-0.0096	-0.0417
10 *	0.0344	0.0287	0.0227	0.0168	0.0110	0.0050	0.0012	-0.0080	-0.0160	-0.0263	-0.0414
11 *	0.0131	0.0238	0.0226	0.0204	0.0173	0.0132	0.0079	0.0012	-0.0064	-0.0158	-0.0392
12 *	0.0162	-0.0076	-0.0069	-0.0069	-0.0071	-0.0072	-0.0073	-0.0072	-0.0059	-0.0034	-0.0036
13 *	-0.0813	-0.0407	-0.0342	-0.0275	-0.0206	-0.0134	-0.0072	-0.0026	-0.0120	-0.0233	0.0467
14 *	-0.0413	-0.0140	-0.0091	-0.0025	0.0028	0.0077	0.0121	0.0151	0.0120	0.0232	0.0660
15 *	0.0836	0.0264	0.0191	0.0120	0.0052	-0.0002	-0.0072	-0.0131	-0.0188	-0.0249	-0.0732
16 *	0.0095	0.0257	0.0193	0.0128	0.0063	-0.0014	-0.0075	-0.0152	-0.0242	-0.0355	-0.0471
17 *	-0.0251	0.0119	0.0105	0.0081	0.0067	0.0040	0.0005	-0.0040	-0.0099	-0.0178	-0.0350
18 *	0.0076	-0.0071	-0.0045	-0.0014	-0.0025	-0.0036	-0.0048	-0.0061	-0.0075	-0.0090	0.0029
19 *	-0.0043	0.0005	-0.0004	0.0009	0.0019	0.0029	0.0038	0.0045	0.0052	0.0057	-0.0121
20 *	0.0060	0.0309	0.0324	0.0340	0.0354	0.0369	0.0384	0.0399	0.0415	0.0431	0.0447
21 *	0.1095	0.0891	0.0881	0.0873	0.0865	0.0858	0.0853	0.0848	0.0845	0.0843	0.1019
22 *	0.1352	0.1197	0.1167	0.1138	0.1111	0.1085	0.1061	0.1038	0.1015	0.0994	0.0984
23 *	0.1114	0.1241	0.1211	0.1182	0.1154	0.1129	0.1104	0.1080	0.1057	0.1033	0.1106
24 *	0.1513	0.1158	0.1117	0.1078	0.1042	0.1007	0.0975	0.0943	0.0912	0.0882	0.0942
25 *	0.0666	0.0836	0.0796	0.0759	0.0725	0.0692	0.0661	0.0631	0.0603	0.0574	0.0415

rcurve array

i	1	2	3	4	5	6	7	8	9	10	11
1 *	0.0001	0.0001	0.0002	0.0002	0.0003	0.0004	0.0005	0.0006	0.0007	0.0009	0.0010
2 *	0.0002	0.0001	0.0001	0.0001	0.0001	0.0001	0.0000	0.0000	-0.0001	-0.0002	0.0020
3 *	0.0001	-0.0001	0.0002	0.0006	0.0010	0.0016	0.0022	0.0030	0.0040	0.0055	-0.0066
4 *	-0.0021	-0.0013	-0.0003	0.0007	0.0018	0.0031	0.0046	0.0065	0.0090	0.0127	0.0432
5 *	-0.0049	0.0022	0.0026	0.0034	0.0044	0.0058	0.0075	0.0099	0.0132	0.0184	-0.0031
6 *	0.0212	-0.0098	-0.0045	-0.0002	0.0039	0.0082	0.0130	0.0187	0.0263	0.0368	0.0762
7 *	-0.1086	-0.0227	-0.0158	-0.0106	-0.0057	-0.0006	0.0048	0.0110	0.0184	0.0275	0.0614
8 *	-0.1046	-0.0400	-0.0277	-0.0200	-0.0136	-0.0077	-0.0018	0.0045	0.0124	0.0218	0.0432
9 *	0.0586	0.0047	0.0037	0.0026	0.0014	0.0001	-0.0014	-0.0033	-0.0060	-0.0096	-0.0417
10 *	0.0344	0.0287	0.0277	0.0168	0.0110	0.0050	-0.0013	-0.0080	-0.0160	-0.0263	-0.0414
11 *	0.0131	0.0238	0.0226	0.0204	0.0173	0.0132	0.0079	0.0012	-0.0064	-0.0158	-0.0392
12 *	0.0162	-0.0076	-0.0069	-0.0069	-0.0071	-0.0072	-0.0073	-0.0073	-0.0059	-0.0034	-0.0036
13 *	-0.0813	-0.0407	-0.0342	-0.0275	-0.0206	-0.0134	-0.0058	0.0026	0.0120	0.0233	0.0467
14 *	-0.0413	-0.0140	-0.0081	-0.0025	0.0028	0.0077	0.0121	0.0160	0.0197	0.0232	0.0660
15 *	0.0836	0.0264	0.0191	0.0120	0.0052	-0.0012	-0.0072	-0.0131	-0.0186	-0.0249	-0.0732
16 *	0.0095	0.0257	0.0193	0.0128	0.0063	-0.0004	-0.0075	-0.0132	-0.0242	-0.0355	-0.0471
17 *	0.0251	0.0119	0.0105	0.0088	0.0065	0.0040	0.0025	-0.0040	-0.0099	-0.0178	-0.0350
18 *	-0.0076	0.0005	-0.0004	-0.0014	-0.0025	-0.0036	-0.0048	-0.0061	-0.0075	-0.0090	0.0029
19 *	-0.0043	-0.0013	-0.0002	0.0009	0.0015	0.0029	0.0038	0.0045	0.0052	0.0057	-0.0121
20 *	0.0060	0.0309	0.0324	0.0340	0.0354	0.0369	0.0384	0.0399	0.0415	0.0431	0.0447
21 *	0.1095	0.0891	0.0881	0.0873	0.0865	0.0858	0.0853	0.0848	0.0845	0.0843	0.1019
22 *	0.1352	0.1197	0.1167	0.1138	0.1111	0.1085	0.1061	0.1038	0.1015	0.0994	0.0884
23 *	0.1214	0.1241	0.1211	0.1182	0.1155	0.1129	0.1104	0.1080	0.1057	0.1033	0.1106
24 *	0.1313	0.1158	0.1117	0.1078	0.1042	0.1007	0.0975	0.0943	0.0912	0.0882	0.0942
25 *	0.0666	0.0836	0.0796	0.0759	0.0725	0.0692	0.0661	0.0631	0.0603	0.0574	0.0415

*** computed compressor parameters for a rotational speed of 16042.8 rpm which is a speed fraction of 1.0000 ***

** the corrected weightflow per unit of casing annular area at the inlet face of the first blade row is 38.90 lbs/sec/ft sq **

** mass averaged rotor and stage aerodynamic parameters **

stage	blade	flow	head	id. head	temp.	adia.	poly.	reset	for. ax.	gas bending moments	torque	power	
no.	type	coef.	coef.	coef.	ratio	ratio	eff.	angle	thrust	for. ax. tang.	(ft-lbs)	(hp)	
								(deg)	(lbs)	(ft-lbs)	(ft-lbs)		
1	rotor	0.4445	0.2412	0.2630	1.6462	1.1605	0.9170	0.9224	0.00	1057.26	17.651	-12.757	703.85
1	stator	0.4148	0.2284	0.2630	1.6061	1.1605	0.8686	0.8768	0.00	-229.24	2.376	6.061	2149.95

TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

2	rotor	0.4607	0.2731	0.2906	1.5527	1.1345	0.9397	0.9431	0.00	1208.74	7.058	-4.819	697.02	2129.08
2	stator	0.4161	0.2607	0.2906	1.5235	1.1346	0.9369	0.9022	0.00	-257.39	1.438	3.614		

** cumulative sums of mass averaged rotor and stage aerodynamic parameters **

stage blade no. type	weight (lbs/sec)	total press. (psia)	total temp. (deg. r.)	press. ratio	temp. ratio	head coef.	ideal head coef.	adla. eff.	poly. eff.	for. ax. thrust (lbs)	torque (ft-lbs)	power (hp)	fract energy
1 inlet	73.30	14.666	518.70										
1 rotor	73.30	24.143	601.97	1.6462	1.1605	0.2412	0.2630	0.9170	0.9224	1057.26	703.85	2149.95	0.5023
1 stator	73.30	23.555	601.97	1.6061	1.1605	0.2284	0.2630	0.8686	0.8768	828.03			
2 rotor	73.30	36.572	682.98	2.4937	1.3167	0.4688	0.5236	0.8954	0.9073	1889.22	1401.16	4279.91	1.0000
2 stator	73.30	35.887	682.98	2.4470	1.3167	0.4579	0.5236	0.8745	0.8885	1631.83			

** values of parameters on streamlines at station, 1, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg. r.)	static press. (psia)	static temp. (deg. r.)
tip 10.099	-11.000	514.40	514.40	0.00	514.40	0.4727	0.00	-0.13	0.000	14.142	518.70	12.151	497.25
1 10.099	-11.000	571.81	571.82	0.00	571.82	0.5281	0.00	-0.15	0.000	14.570	518.70	12.150	492.18
2 9.632	-11.000	574.77	574.78	0.00	574.78	0.5310	0.00	-0.22	0.000	14.700	518.70	12.151	491.50
3 9.167	-11.000	574.69	574.69	0.00	574.69	0.5309	0.00	-0.30	0.000	14.700	518.70	12.152	491.91
4 8.675	-11.000	574.58	574.58	0.00	574.58	0.5308	0.00	-0.38	0.000	14.700	518.70	12.152	491.92
5 8.154	-11.000	574.43	574.43	0.00	574.43	0.5306	0.00	-0.48	0.000	14.700	518.70	12.152	491.93
6 7.597	-11.000	574.23	574.23	0.00	574.23	0.5305	0.00	-0.60	0.001	14.700	518.70	12.155	491.95
7 6.956	-11.000	573.59	573.59	0.00	573.59	0.5302	0.00	-0.75	0.001	14.700	518.70	12.157	491.97
8 6.338	-11.000	573.64	573.72	0.00	573.72	0.5300	0.00	-0.94	0.001	14.700	518.70	12.160	492.00
9 5.693	-11.000	573.30	573.43	0.00	573.43	0.5297	0.00	-1.23	0.000	14.700	518.70	12.162	492.03
10 4.755	-11.000	568.59	568.88	0.00	568.88	0.5253	0.00	-1.84	0.001	14.660	518.70	12.166	492.45
11 3.714	-11.000												
Hub 3.714	-11.000												

** values of parameters on streamlines at station, 2, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg. r.)	static press. (psia)	static temp. (deg. r.)
tip 10.099	-9.000	516.05	516.05	0.00	516.05	0.4743	0.00	-0.11	0.000	14.178	518.70	12.170	497.11
1 10.095	-9.000	569.34	569.34	0.00	569.34	0.5257	0.00	-0.13	0.000	14.569	518.70	12.170	492.41
2 9.627	-9.000	572.19	572.19	0.00	572.19	0.5285	0.00	-0.18	0.000	14.698	518.70	12.170	492.14
3 9.160	-9.000	572.03	572.03	0.00	572.03	0.5283	0.00	-0.24	0.001	14.698	518.70	12.172	492.16
4 8.666	-9.000	571.80	571.81	0.00	571.81	0.5281	0.00	-0.31	0.001	14.698	518.70	12.174	492.18
5 8.142	-9.000	571.50	571.52	0.00	571.52	0.5278	0.00	-0.40	0.001	14.698	518.70	12.176	492.20
6 7.582	-9.000	571.12	571.15	0.00	571.15	0.5275	0.00	-0.51	0.001	14.698	518.70	12.179	492.24
7 6.977	-9.000	570.66	570.69	0.00	570.69	0.5270	0.00	-0.65	0.001	14.698	518.70	12.183	492.28
8 6.314	-9.000												

TABLE 2 - EXAMPLE OUTPUT DATA SET (no statior reset)

9	5.572	-9.000	570.11	570.18	0.00	570.18	0.5265	0.00	-0.85	0.001	14.698	518.70	12.187	492.33
10	4.713	-9.000	569.52	569.64	0.00	569.64	0.5260	0.00	-1.15	0.001	14.698	518.70	12.192	492.38
11	3.655	-9.000	564.50	564.71	0.00	564.71	0.5212	0.00	-1.54	0.002	14.658	518.70	12.199	492.83
hub	3.643	-9.000												

1

** values of parameters on streamlines at station, 3, which is an annulus **

streamline no. radius	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.f.)	static press. (psia)	static temp. (deg.f.)
tip 10.100	-7.000	517.54	517.54	0.00	517.54	0.4758	0.00	-0.09	0.000	14.209	518.70	12.186	496.98
1	10.091	-7.000	567.28	0.00	567.28	0.5237	0.00	-0.08	0.000	14.668	518.70	12.186	492.60
2	9.624	-7.000	570.02	0.00	570.02	0.5261	0.00	-0.11	0.001	14.696	518.70	12.187	492.34
3	9.154	-7.000	569.76	0.00	569.76	0.5261	0.00	-0.14	0.001	14.696	518.70	12.189	492.37
4	8.659	-7.000	569.35	0.00	569.35	0.5257	0.00	-0.18	0.002	14.696	518.70	12.192	492.40
5	8.134	-7.000	568.75	0.00	568.75	0.5251	0.00	-0.23	0.002	14.696	518.70	12.197	492.46
6	7.571	-7.000	567.89	0.00	567.89	0.5243	0.00	-0.30	0.003	14.696	518.70	12.204	492.54
7	6.963	-7.000	566.67	0.00	566.67	0.5231	0.00	-0.41	0.003	14.696	518.70	12.214	492.65
8	6.295	-7.000	565.10	0.00	565.10	0.5216	0.00	-0.60	0.004	14.696	518.70	12.227	492.80
9	5.546	-7.000	563.25	0.00	563.25	0.5198	0.00	-0.98	0.003	14.696	518.70	12.242	492.97
10	4.676	-7.000	559.87	0.00	559.87	0.5165	0.00	-2.11	-0.007	14.656	518.70	12.237	493.28
11	3.597	-7.000											
hub 3.572	-7.000												

** values of parameters on streamlines at station, 4, which is an annulus **

streamline no. radius	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.f.)	static press. (psia)	static temp. (deg.f.)
tip 10.101	-5.200	520.17	520.17	0.00	520.17	0.4783	0.00	-0.16	-0.002	14.234	518.70	12.187	496.76
1	10.088	-5.200	567.49	0.00	567.49	0.5239	0.00	-0.13	-0.001	14.667	518.70	12.183	492.58
2	9.621	-5.200	570.56	0.00	570.56	0.5269	0.00	-0.09	0.000	14.695	518.70	12.181	492.29
3	9.153	-5.200	569.47	0.00	569.47	0.5268	0.00	-0.04	0.001	14.695	518.70	12.182	492.30
4	8.657	-5.200	569.38	0.00	569.38	0.5263	0.00	0.03	0.002	14.695	518.70	12.186	492.35
5	8.131	-5.200	569.00	0.00	569.00	0.5254	0.00	0.09	0.004	14.695	518.70	12.194	492.44
6	7.568	-5.200	567.39	0.00	567.39	0.5238	0.00	0.14	0.005	14.695	518.70	12.207	492.59
7	6.959	-5.200	564.90	0.00	564.90	0.5214	0.00	0.17	0.008	14.695	518.70	12.228	492.82
8	6.290	-5.200	560.98	0.00	560.98	0.5176	0.00	0.30	0.011	14.695	518.70	12.260	493.17
9	5.537	-5.200	554.29	0.00	554.29	0.5111	0.00	0.07	0.017	14.695	518.70	12.314	493.78
10	4.658	-5.200	532.01	0.00	532.01	0.4896	0.00	0.02	0.043	14.655	518.70	12.457	495.75
11	3.546	-5.200											
hub 3.507	-5.200												

1

TABLE 2 - EXAMPLE OUTPUT DATA SET(no statior reset)

** values of parameters on streamlines at station, 5, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.f.)	static press. (psia)	static temp. (deg.f.)
tip 10.101	-3.700	527.31	527.34	0.00	527.34	0.4852	0.00	-0.65	-0.005	14.253	518.70	12.150	496.15
1 10.079	-3.700	572.30	572.31	0.00	572.31	0.5286	0.00	-0.29	0.001	14.666	518.70	12.143	492.13
2 9.616	-3.700	575.09	575.10	0.00	575.10	0.5313	0.00	-0.09	0.001	14.694	518.70	12.143	491.87
3 9.149	-3.700	574.82	574.82	0.00	574.82	0.5310	0.00	0.08	0.001	14.694	518.70	12.145	491.90
4 8.657	-3.700	574.22	574.23	0.00	574.23	0.5305	0.00	0.25	0.003	14.694	518.70	12.150	491.95
5 8.134	-3.700	573.10	573.11	0.00	573.11	0.5294	0.00	0.45	0.004	14.694	518.70	12.159	492.06
6 7.575	-3.700	571.07	571.11	0.00	571.11	0.5274	0.00	0.69	0.007	14.694	518.70	12.176	492.24
7 6.970	-3.700	567.60	567.68	0.00	567.68	0.5241	0.00	0.97	0.011	14.694	518.70	12.204	492.56
8 6.305	-3.700	561.94	562.09	0.00	562.09	0.5187	0.00	1.30	0.015	14.694	518.70	12.250	493.07
9 5.956	-3.700	553.23	553.46	0.00	553.46	0.5103	0.00	1.64	0.020	14.694	518.70	12.320	493.86
10 4.681	-3.700	544.15	544.37	0.00	544.37	0.5015	0.00	1.64	-0.003	14.654	518.70	12.359	494.67
11 3.581	-3.700												
hub 3.518	-3.700												

** values of parameters on streamlines at station, 6, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.f.)	static press. (psia)	static temp. (deg.f.)
tip 10.099	-2.300	538.38	538.41	0.00	538.41	0.4958	0.00	0.59	0.021	14.269	518.70	12.079	495.19
1 10.070	-2.301	577.50	577.50	0.00	577.50	0.5326	0.00	0.11	0.002	14.666	518.70	12.100	491.64
2 9.611	-2.322	580.54	580.54	0.00	580.54	0.5366	0.00	-0.02	-0.004	14.693	518.70	12.097	491.36
3 9.147	-2.344	581.80	581.80	0.00	581.80	0.5378	0.00	0.06	-0.005	14.693	518.70	12.087	491.24
4 8.658	-2.366	582.87	582.88	0.00	582.88	0.5389	0.00	0.30	-0.003	14.693	518.70	12.078	491.14
5 8.140	-2.390	582.92	582.92	0.00	582.92	0.5389	0.00	0.69	0.001	14.693	518.70	12.077	491.13
6 7.588	-2.415	581.06	581.16	0.00	581.16	0.5372	0.00	1.20	0.007	14.693	518.70	12.092	491.30
7 6.991	-2.442	576.83	576.83	0.00	576.83	0.5330	0.00	1.83	0.014	14.693	518.70	12.128	491.71
8 6.335	-2.472	568.65	568.65	0.00	568.65	0.5250	0.00	2.57	0.022	14.693	518.70	12.195	492.47
9 5.937	-2.506	554.36	554.36	0.00	554.36	0.5112	0.00	3.37	0.034	14.693	518.70	12.311	493.77
10 4.730	-2.546	517.64	517.64	0.00	517.64	0.4758	0.00	3.79	0.076	14.653	518.70	12.566	496.97
11 3.623	-2.596												
hub 3.540	-2.600												

** values of parameters on streamlines at station, 7, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.f.)	static press. (psia)	static temp. (deg.f.)
tip 10.100	-1.000	503.32	504.16	0.00	504.16	0.4629	0.00	-3.30	-0.109	14.282	518.70	12.347	498.09
1 10.065	-1.003	567.73	568.38	0.00	568.38	0.5248	0.00	-2.75	-0.066	14.665	518.70	12.175	492.49
2 9.594	-1.039	585.24	585.66	0.00	585.66	0.5416	0.00	-2.15	-0.044	14.692	518.70	12.054	490.87
3 9.129	-1.075	595.48	595.67	0.00	595.67	0.5514	0.00	-1.45	-0.029	14.692	518.70	11.969	489.91
4 8.647	-1.113	601.95	601.98	0.00	601.98	0.5575	0.00	-0.60	-0.017	14.692	518.70	11.916	489.30
5 8.139	-1.152												

TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

streamline no. radius (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg. r.)	static press. (psia)	static temp. (deg. r.)
tip 10.034	0.137	583.34	584.62	0.00	584.62	0.5406	0.00	-3.79	-0.105	14.284	518.70
1 9.913	0.164	583.04	637.15	0.00	637.15	0.5222	0.00	-6.51	-0.086	14.665	518.70
2 9.490	-0.009	641.47	643.60	0.00	643.60	0.5396	0.00	-4.66	-0.018	14.891	518.70
3 9.055	-0.008	645.10	646.04	0.00	646.04	0.5610	0.00	-3.09	-0.012	14.891	518.70
4 8.599	-0.072	645.55	645.79	0.00	645.79	0.6008	0.00	-1.57	-0.007	14.691	518.70
5 8.117	-0.146	643.05	643.06	0.00	643.06	0.5981	0.00	0.06	-0.002	14.691	518.70
6 7.601	-0.228	637.10	637.41	0.00	637.41	0.5925	0.00	1.80	0.003	14.691	518.70
7 7.043	-0.322	627.22	628.49	0.00	628.49	0.5836	0.00	3.65	0.010	14.691	518.70
8 6.428	-0.435	612.28	615.33	0.00	615.33	0.5707	0.00	5.70	0.018	14.691	518.70
9 5.733	-0.561	589.87	595.80	0.00	595.80	0.5515	0.00	8.09	0.029	14.691	518.70
10 4.914	-0.722	557.25	569.33	0.00	569.33	0.5257	0.00	11.82	0.043	14.652	518.70
11 3.883	-0.879										
hub 3.762	-0.898										

** values of parameters on streamlines at station, 8, which is the inlet of rotor number, 1 **

streamline no. r/rip	rel. flow angle (deg)	rel. vel. tang. vel. (ft/sec)	rel. mach number	wheel speed (ft/sec)	flow coef.
tip 1.0000	67.16	1387.87	1.3925	1387.87	0.4152
1 0.9879	64.38	1328.58	1.3695	1328.58	0.4506
2 0.9457	63.08	1267.74	1.3223	1267.74	0.4566
3 0.9024	61.78	1203.82	1.2710	1203.82	0.4592
4 0.8569	60.39	1136.32	1.2159	1136.32	0.4595
5 0.8089	58.86	1064.19	1.1564	1064.19	0.4578
6 0.7575	57.12	986.07	1.0913	986.07	0.4535
7 0.7019	55.07	899.95	1.0193	899.95	0.4465
8 0.6406	52.52	802.57	0.9379	802.57	0.4358
9 0.5713	49.11	688.02	0.8425	688.02	0.4199
10 0.4898	43.68	543.68	0.7269	543.68	0.3967
11 0.3870					
hub 0.3749					

streamline no. pect. pass.	inc. angle (deg)	ref. angle (deg)	s.s. inc. angle (deg)	in. blade angle (deg)	streamline element sh. loc. cov. chan. min. chb. pt. loc. in area margin cov. chan.
1 1.93	1.19	-2.25	-0.92	65.97	-0.0019
2 6.68	1.24	-2.08	-0.94	63.14	0.2135
3 15.61	1.87	-1.96	-0.53	62.22	0.3959
4 22.89	2.41	-1.88	-0.33	59.37	0.4483
5 30.57	3.12	-1.86	-0.09	57.27	0.4800
6 38.79	3.88	-1.86	0.03	54.98	0.4527
7 47.68	4.76	-1.89	0.13	52.36	0.4046
8 57.49	5.72	-1.92	0.17	49.35	0.3658
9 68.58	6.76	-1.97	-0.03	45.76	0.2718
10 81.62	7.12	-2.13	-0.37	41.99	0.2730
11 98.06	6.15	-2.57	-1.66	37.53	0.2540
					0.2870

** values of parameters on streamlines at station, 9, which is the outlet of rotor number, 1 **

TABLE 2 - EXAMPLE OUTPUT DATA SET (no statior reset)

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. val. (ft/sec)	abs. mach no.	abs. angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)	
tip	9.756	1.778	525.38	530.08	460.98	702.48	0.6001	41.01	-7.63	0.059	24.679	618.54	19.412	579.29
1	9.661	1.793	535.09	537.32	438.42	693.49	0.5961	39.21	-5.22	0.042	24.587	610.08	19.397	571.76
2	9.288	1.852	519.13	520.23	430.07	674.98	0.5818	39.58	-3.73	0.030	24.175	604.67	19.279	568.33
3	8.903	1.887	517.84	518.26	436.02	677.28	0.5883	40.07	-2.30	0.022	24.053	601.86	19.130	565.25
4	8.492	1.943	522.04	522.10	449.18	688.73	0.5968	40.71	-0.87	0.016	24.025	600.04	18.940	562.16
5	8.061	2.012	531.22	531.25	476.79	713.83	0.6199	41.91	0.62	0.010	24.147	600.22	18.693	559.52
6	7.611	2.105	533.93	533.91	504.80	734.77	0.6396	43.39	2.16	0.002	24.124	599.64	18.377	556.49
7	7.137	2.208	537.45	538.65	542.54	764.52	0.6678	45.21	3.83	-0.006	24.132	599.49	17.956	552.76
8	6.628	2.331	543.82	543.82	583.81	797.85	0.7001	47.03	5.70	-0.015	24.025	598.39	17.380	547.47
9	6.075	2.477	546.02	546.02	642.44	846.65	0.7478	49.36	8.04	-0.026	23.912	597.47	16.562	540.07
10	5.456	2.609	554.02	554.02	737.20	929.22	0.8298	52.50	11.38	-0.042	23.832	597.02	15.232	537.79
11	4.728	2.703												
hub	4.664	2.712												
streamline no. r/rtip	rel. flow angle (deg)	rel. tang. vel. (ft/sec)	rel. mach number	rel. mach speed (ft/sec)	flow coeff.	head coeff.	ideal head coeff.	adiab. eff.	diffusion factor	loss coeff.	shock loss coeff.	shock solidity		
tip	1.0000													
1	0.9902	891.51	1037.20	0.8860	1332.49	0.3740	0.2560	0.3159	0.8419	0.4271	0.1262	0.0439	1.3046	
2	0.9521	58.06	861.97	1015.73	0.8730	1300.39	0.3809	0.2507	0.2889	0.4186	0.1012	0.0390	1.3630	
3	0.9126	57.49	816.33	968.00	0.8344	1246.40	0.3695	0.2410	0.2716	0.4244	0.0859	0.0331	1.4253	
4	0.8704	55.46	752.83	913.97	0.7899	1186.84	0.3686	0.2383	0.2627	0.4369	0.0721	0.0279	1.4979	
5	0.8263	52.46	679.34	856.79	0.7424	1128.52	0.3716	0.2377	0.2569	0.4527	0.0603	0.0231	1.5826	
6	0.7802	47.94	588.78	793.02	0.6887	1065.57	0.3781	0.2404	0.2574	0.4762	0.0576	0.0191	1.6812	
7	0.7315	42.80	494.37	727.64	0.6336	999.17	0.3798	0.2399	0.2556	0.5001	0.0575	0.0149	1.8065	
8	0.6794	35.59	385.43	662.35	0.5786	927.96	0.3826	0.2400	0.2551	0.5244	0.0608	0.0118	1.9636	
9	0.6227	26.13	266.74	605.71	0.5315	850.54	0.3852	0.2377	0.2516	0.5375	0.0638	0.0082	2.1763	
10	0.5593	12.42	121.41	564.65	0.4987	763.84	0.3887	0.2353	0.2487	0.5286	0.0729	0.0001	2.4929	
11	0.4846	-7.58	-75.32	570.66	0.5096	661.88	0.3947	0.2350	0.2472	0.5503	0.0859	0.0000	3.0528	
hub	0.4780													
streamline no. pct. pass.	press. ratio	temp. ratio	local blade forces radius for axial (in.)	tang. (lbs/in)	dev. angle (deg)	out. stream. angle (deg)								
1	1.87	1.7265	9.7870	20.0553	-11.9990	5.26	54.01							
2	9.18	1.6786	9.3892	18.5167	-11.2417	4.95	53.11							
3	16.75	1.6455	8.9791	16.8884	-10.5630	4.66	52.83							
4	24.83	1.6372	8.5452	15.6417	-10.1490	4.71	50.75							
5	33.29	1.6353	8.0887	14.4177	-9.9156	4.92	47.53							
6	42.12	1.6436	7.6063	13.1753	-9.9002	5.49	42.45							
7	51.43	1.6420	7.0902	11.6547	-9.7158	6.18	36.62							
8	61.42	1.6426	6.5283	9.9691	-9.4993	7.25	28.33							
9	72.28	1.6353	5.9040	8.0061	-9.0478	8.61	17.52							
10	84.44	1.6276	5.1519	5.6451	-8.4096	10.49	1.92							
11	98.74	1.6265	4.3056	2.6711	-8.6788	11.84	-19.42							

** values of parameters on streamlines at station, 10, which is an annulus **

TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

streamline no	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. angle (deg)	flow stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	9.630	3.000	561.33	561.86	465.12	729.40	0.6251	39.62	-2.50	0.034	24.669	617.67	19.024	575.32
1	9.575	3.025	574.95	575.30	441.38	725.11	0.6253	37.50	-2.01	0.029	24.580	609.83	18.950	567.91
2	9.226	3.047	558.21	558.38	432.04	706.01	0.6103	37.73	-1.40	0.024	24.175	604.67	18.859	564.90
3	8.862	3.071	555.06	555.06	436.90	706.38	0.6121	38.21	-0.61	0.018	24.055	601.93	18.738	562.09
4	8.474	3.096	556.48	556.48	448.82	714.92	0.6210	38.69	0.29	0.013	24.027	600.14	18.584	559.30
5	8.067	3.122	562.34	562.47	475.01	736.21	0.6407	40.18	1.23	0.008	24.148	600.32	18.381	557.01
6	7.640	3.150	562.11	562.55	501.35	753.53	0.6573	41.71	2.27	0.002	24.135	599.73	18.112	554.35
7	7.186	3.180	563.16	564.20	537.02	778.92	0.6814	43.59	3.48	-0.004	24.133	599.59	17.749	551.07
8	6.697	3.213	563.37	565.47	575.77	807.01	0.7088	45.52	4.93	-0.011	24.026	598.49	17.247	546.38
9	6.160	3.250	565.13	569.14	630.52	849.40	0.7504	47.93	6.80	-0.022	23.914	597.57	16.522	539.79
10	5.559	3.293	567.77	577.54	717.09	920.74	0.8212	51.15	10.55	-0.041	23.834	597.13	15.367	529.16
hub	4.750	3.300												

1 ** values of parameters on streamlines at station, 11, which is the inlet of stator number, 1, of stage number, 1 **

streamline no.	radius coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)	
tip	9.599	4.118	578.27	578.35	466.89	743.29	0.6382	38.91	-0.97	0.013	24.660	616.95	18.814	572.96
1	9.538	4.117	578.27	578.35	466.89	743.29	0.6382	38.91	-0.97	0.013	24.660	616.95	18.814	572.96
2	9.197	4.115	594.55	594.57	442.77	741.32	0.6404	36.67	-0.45	0.017	24.573	609.60	18.713	565.78
3	8.843	4.119	579.99	579.89	433.00	723.72	0.6267	36.75	0.03	0.017	24.174	604.67	18.614	562.86
4	8.466	4.123	576.92	576.95	437.34	723.97	0.6284	37.16	0.54	0.014	24.056	601.99	18.497	560.13
5	8.071	4.128	578.13	578.23	448.64	731.87	0.6367	37.81	1.08	0.010	24.028	600.21	18.350	557.42
6	7.655	4.135	583.61	583.87	474.04	752.08	0.6556	39.07	1.70	0.006	24.150	600.40	18.157	555.19
7	7.215	4.143	583.59	583.59	499.34	768.06	0.6710	40.55	2.45	0.002	24.126	599.81	17.904	552.65
8	6.740	4.151	583.41	584.40	533.53	791.32	0.6931	42.39	3.35	-0.001	24.134	599.67	17.568	549.59
9	6.221	4.161	582.38	584.13	570.14	816.26	0.7177	44.31	4.44	-0.006	24.028	598.58	17.111	545.26
10	5.640	4.173	582.71	585.61	621.47	853.91	0.7548	46.70	5.71	-0.015	23.916	597.66	16.456	539.26
11	4.969	4.180	589.30	594.04	701.35	919.12	0.8196	49.74	7.25	-0.039	23.836	597.22	15.395	539.49
hub	4.847	4.181												

streamline no. r/rtip	flow coef.	rel flow angle (deg)
tip 1.0000	0.4116	56.34
1	0.9937	54.86
2	0.9581	54.23
3	0.9212	52.35
4	0.8820	49.68
5	0.8408	45.67
6	0.7975	41.19
7	0.7516	35.06
8	0.7022	27.24
9	0.6481	16.02
10	0.5876	-0.54
11	0.5177	
hub 0.5050		

streamline				inlet streamline			streamline element				
no.	pct. pass.	inc. angle (deg)	ref. s.s. angle (deg)	inc. angle (deg)	in. blade angle (deg)	element slope (deg)	mach no. at shock location	sh. loc. as fract of s.s.	cov. chan. margin	min. chk. pt. loc. in cov. chan.	
1	1.28	2.18	-2.25	-3.99	36.73	-0.30	0.8593	0.3342	0.5159	0.1942	0.0000
2	8.45	3.13	-2.08	-3.04	33.54	0.27	0.8318	0.3097	0.5727	0.1761	0.0000
3	15.92	3.34	-1.96	-2.80	33.41	0.67	0.8057	0.3026	0.5989	0.1905	0.0000
4	23.84	3.32	-1.88	-2.78	33.84	1.00	0.8033	0.2967	0.6153	0.1888	0.0000

TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

5	32.17	3.27	-1.86	-2.78	34.54	1.35	0.8106	0.2911	0.6293	0.1799	0.0000	0.0000	0.0000	0.0000
6	40.90	3.32	-1.86	-2.68	35.76	1.82	0.8359	0.2872	0.6393	0.1656	0.0000	0.0000	0.0000	0.0000
7	50.17	3.36	-1.89	-2.57	37.19	2.45	0.8573	0.2824	0.6493	0.1573	0.0000	0.0000	0.0000	0.0000
8	60.16	3.55	-1.92	-2.31	38.84	3.21	0.8898	0.2759	0.6583	0.1495	0.0000	0.0000	0.0000	0.0000
9	71.09	3.56	-1.97	-2.23	40.75	4.10	0.9226	0.2674	0.6598	0.1409	0.0000	0.0000	0.0000	0.0000
10	83.31	3.30	-2.13	-2.38	43.40	4.99	0.9723	0.2568	0.6777	0.1300	0.0000	0.0000	0.0000	0.0000
11	97.43	2.86	-2.57	-2.67	46.87	5.41	1.0608	0.2402	0.6804	0.1134	0.0000	0.0000	0.0000	0.0000

1

** values of parameters on streamlines at station, 12, which is the outlet of stator number, 1, of stage number, 1 **

streamline no.	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. angle (deg)	abs. flow stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	9.596	6.353	601.43	601.49	-44.71	603.15	0.5131	-4.25	0.82	0.016	24.147	613.29	20.231	584.35
1	8.526	6.352	594.78	594.84	-43.36	595.30	0.5081	-2.25	0.79	0.002	24.094	608.37	20.252	580.15
2	7.508	6.347	587.57	587.62	-42.07	587.71	0.4851	-1.02	0.80	-0.006	23.731	604.58	20.250	578.90
3	6.869	6.345	579.73	579.80	-40.88	579.82	0.4790	-0.50	0.93	-0.008	23.623	602.26	20.235	577.28
4	6.305	6.344	559.40	559.52	-41.26	559.53	0.4785	-0.13	1.22	-0.008	23.595	600.63	20.218	575.76
5	6.123	6.344	538.58	538.68	-40.98	538.68	0.4866	0.10	1.64	-0.007	23.697	600.81	20.201	575.11
6	7.726	6.343	567.44	567.51	-40.98	567.51	0.4866	0.10	1.64	-0.007	23.697	600.81	20.201	575.11
7	7.309	6.343	562.91	563.32	-40.98	563.33	0.4796	0.27	2.20	-0.007	23.624	600.24	20.185	574.33
8	6.863	6.344	558.80	559.51	-40.98	559.51	0.4829	0.13	2.87	-0.007	23.555	600.09	20.182	573.58
9	6.377	6.345	543.60	544.68	-40.98	544.68	0.4668	-0.02	3.61	-0.008	23.347	599.04	20.152	573.38
10	5.830	6.347	518.00	519.45	-40.98	519.53	0.4447	-0.01	4.28	-0.008	23.042	596.15	20.129	576.62
11	5.175	6.355	468.52	469.96	-40.98	471.01	0.4021	-3.82	4.49	-0.004	22.436	597.72	20.107	580.03
hub	5.043	6.356												

streamline no.	r/rtip	flow coef.	head coef.	ideal head coef.	stator po.ratio	stage po.ratio	stage ad.eff.	diffusion factor	loss coef.	shock loss	element solidity
tip	1.0000	0.4281	0.2546	0.3159	0.9792	1.6893	0.8059	0.4555	0.0876	0.0000	1.2900
1	0.9927	0.4281	0.2546	0.3159	0.9792	1.6893	0.8059	0.4555	0.0876	0.0000	1.2900
2	0.9595	0.4234	0.2401	0.2889	0.9805	1.6429	0.8312	0.4322	0.0818	0.0000	1.3361
3	0.9241	0.4040	0.2314	0.2716	0.9817	1.6153	0.8519	0.4357	0.0797	0.0000	1.3885
4	0.8862	0.3984	0.2290	0.2627	0.9820	1.6079	0.8720	0.4370	0.0780	0.0000	1.4493
5	0.8464	0.3975	0.2284	0.2569	0.9820	1.6060	0.8893	0.4385	0.0763	0.0000	1.5191
6	0.8050	0.4039	0.2306	0.2574	0.9813	1.6130	0.8959	0.4409	0.0755	0.0000	1.5997
7	0.7616	0.4007	0.2290	0.2556	0.9792	1.6080	0.8962	0.4561	0.0808	0.0000	1.5947
8	0.7152	0.3978	0.2276	0.2551	0.9760	1.6033	0.8920	0.4770	0.0882	0.0000	1.8104
9	0.6646	0.3870	0.2230	0.2516	0.9716	1.5891	0.8862	0.5091	0.0985	0.0000	1.9565
10	0.6075	0.3687	0.2156	0.2487	0.9622	1.5663	0.8670	0.5605	0.1213	0.0000	2.1509
11	0.5393	0.3335	0.2041	0.2472	0.9414	1.5314	0.8256	0.6484	0.1656	0.0007	2.4327
hub	0.5256										

streamline no.	pt. pass.	local blade forces		radius for axial (in.)	tang. (lbs/in.)	dev. angle (deg)	out. blade angle (deg)
1	1.54	9.5357	2.9684	10.3937	11.93	-16.18	10.21
2	8.54	9.2025	2.6194	9.2374	10.21	-12.46	9.58
3	15.99	8.8556	2.4000	8.2889	9.58	-10.60	9.24
4	23.98	8.4854	2.4109	7.8124	9.24	-9.74	8.99
5	32.37	8.0966	2.4442	7.5881	8.99	-9.12	8.90
6	41.09	7.6904	2.4172	7.6009	8.90	-8.81	8.90
7	50.24	7.2619	2.7153	7.4989	8.90	-8.81	8.90
8	60.03	6.8016	2.8632	7.4237	8.91	-8.79	8.91
9	70.70	6.2992	2.9580	7.1902	8.96	-8.98	8.96
10	82.73	5.7349	3.0462	6.9337	9.27	-10.28	9.27
11	97.10	5.0724	3.0197	7.4102	10.39	-14.21	10.39

TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

** values of parameters on streamlines at station, 13, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip 9.603	7.340	578.89	579.01	-44.68	580.73	0.4933	-4.41	-1.16	-0.081	24.139	612.98	20.491	586.15
1 9.534	7.340	587.79	587.92	-23.35	588.39	0.5020	-2.27	-1.20	-0.085	24.087	608.24	20.328	580.67
2 9.210	7.340	597.79	597.92	-10.06	597.91	0.4888	-1.01	-0.92	-0.047	23.729	604.56	20.199	578.49
3 8.871	7.340	571.15	571.82	-4.87	571.19	0.4891	-0.49	-0.43	-0.033	23.622	602.28	20.104	576.27
4 8.511	7.340	574.28	574.28	-1.26	574.28	0.4925	-0.13	0.21	-0.022	23.595	600.67	20.036	574.37
5 8.137	7.340	585.52	585.60	0.98	585.60	0.5026	0.10	0.94	-0.013	23.697	600.85	19.990	573.50
6 7.749	7.340	581.85	582.12	2.62	582.13	0.4997	0.26	1.75	-0.006	23.624	600.28	19.966	573.25
7 7.344	7.340	576.59	577.21	1.22	577.21	0.4954	0.12	2.66	0.002	23.595	600.13	19.965	573.56
8 6.311	7.340	557.94	559.08	-0.23	559.08	0.4796	-0.02	3.65	0.011	23.348	599.09	19.992	574.15
9 5.906	7.340	524.75	526.57	-9.08	526.65	0.4510	-0.99	4.77	0.026	23.017	598.20	20.059	576.08
10 5.397	7.340	461.30	463.66	-30.88	464.68	0.3965	-3.81	5.78	0.047	22.453	597.77	20.180	580.55
11 5.131	7.340												
hub 5.131	7.340												

** values of parameters on streamlines at station, 14, which is the inlet of rotor number, 2 **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip 9.499	8.707	645.23	649.68	-45.17	651.25	0.5565	-3.98	-6.71	-0.041	24.128	612.56	19.609	578.79
1 9.430	8.703	650.44	653.21	-23.57	653.63	0.5608	-2.07	-5.27	-0.027	24.077	608.07	19.506	574.02
2 9.128	8.686	632.72	634.17	-10.13	634.25	0.5448	-0.92	-3.87	-0.020	23.727	604.53	19.443	572.45
3 8.812	8.664	628.51	629.06	-4.89	629.08	0.5412	-0.45	-2.41	-0.014	23.622	602.31	19.407	570.74
4 8.477	8.638	627.24	627.33	-1.26	627.33	0.5403	-0.12	-0.96	-0.007	23.594	600.73	19.396	569.32
5 8.127	8.609	633.21	633.23	0.98	633.23	0.5456	0.09	0.41	-0.001	23.696	600.90	19.407	568.90
6 7.762	8.578	625.39	625.68	2.60	625.68	0.5390	0.24	1.73	0.003	23.624	600.33	19.438	569.09
7 7.379	8.543	615.96	616.88	1.21	616.88	0.5311	0.11	3.13	0.009	23.555	600.19	19.490	569.83
8 6.969	8.504	593.50	595.54	-0.22	595.54	0.5123	-0.02	4.74	0.016	23.351	599.15	19.571	570.85
9 6.521	8.462	556.00	559.96	-8.91	560.03	0.4808	-0.91	6.81	0.028	23.024	598.27	19.699	573.25
10 6.014	8.412	490.54	490.54	-30.09	491.46	0.4200	-3.51	9.71	0.066	22.469	597.84	19.936	578.58
11 5.397	8.337												
hub 5.265	8.321												

streamline no. r/rtip	rel. flow angle (deg)	rel. tang. vel. (ft/sec)	rel. mach number	wheel speed (ft/sec)	flow coef.
tip 1.0000					
1 0.9927	64.55	1365.35	1.2921	1329.87	0.4852
2 0.9610	63.38	1301.51	1.2493	1277.95	0.4891
3 0.9277	62.96	1243.79	1.1993	1233.66	0.4758
4 0.8924	62.17	1191.64	1.1592	1186.75	0.4726
5 0.8555	61.15	1138.97	1.1199	1137.71	0.4717
6 0.8171	59.75	1085.70	1.0829	1086.68	0.4761
7 0.7768	58.73	1030.45	1.0385	1033.06	0.4703
8 0.7336	57.66	974.39	0.9928	975.60	0.4632
9 0.6865	56.89	913.13	0.9377	912.90	0.4463
10 0.6331	56.65	850.81	0.8744	841.89	0.4181
11 0.5682	58.02	785.70	0.7917	755.61	0.3636
hub 0.5543					

TABLE 2 - EXAMPLE OUTPUT DATA SET (no statior reset)

inlet streamline				streamline element			
streamline no.	inc. angle (deg)	ref. s.s. inc. angle (deg)	inc. angle (deg)	element mach no. slope at shock as fract of s.s. location (deg)	sh. loc. cov. chan. min. chk. area pt. loc. in cov. chan.	min. chk. area pt. loc. in cov. chan.	
1	1.63	2.52	-2.25	-0.02	62.03	-6.58	1.4015
2	8.76	2.24	-2.08	-0.26	61.11	-5.08	1.3534
3	16.23	2.96	-1.96	0.36	60.02	-3.75	1.3227
4	24.14	3.43	-1.88	0.58	58.74	-2.32	1.2989
5	32.42	3.85	-1.86	0.64	57.31	-0.88	1.2774
6	41.03	3.98	-1.86	0.31	55.77	0.48	1.2549
7	50.07	4.53	-1.89	0.34	54.20	1.79	1.2386
8	59.76	5.02	-1.92	0.30	52.64	3.23	1.2242
9	70.34	5.70	-1.97	0.49	51.19	4.93	1.1806
10	82.32	6.20	-2.13	0.68	50.45	7.22	1.1325
11	96.88	5.22	-2.57	-0.09	52.80	10.98	1.0776
** values of parameter on streamlines at station, 15, which is the outlet of rotor number, 2 **							
streamline no.	radius (in.)	axial coord. (in.)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. mach no. (deg)	abs. flow stream. slope (deg)	total stream. curv. (1./in.)
tip	9.374	9.729	546.69	548.82	365.30	659.28	0.5283
1	9.311	9.733	526.28	527.50	384.77	652.92	0.5258
2	9.034	9.750	506.62	506.62	419.50	662.19	0.5344
3	8.739	9.774	511.70	512.36	439.02	670.56	0.5425
4	8.430	9.801	506.62	506.62	454.57	680.57	0.5521
5	8.108	9.832	506.47	507.57	470.49	690.11	0.5623
6	7.773	9.867	507.47	507.47	495.05	708.41	0.5765
7	7.422	9.908	506.49	506.49	522.76	728.16	0.5934
8	7.050	9.954	506.18	506.18	557.68	751.79	0.6140
9	6.624	10.007	502.58	504.16	605.59	785.43	0.6427
10	6.224	10.069	497.03	500.16	659.32	829.23	0.7042
11	5.747	10.143	494.10	500.63	698.32	859.23	0.7442
hub	5.647	10.158					
streamline no.	r/rtip	rel. flow angle (deg)	rel. tang. vel. (ft/sec)	rel. merid. vel. (ft/sec)	rel. number	wheel speed (ft/sec)	flow coeff.
tip	1.0000	59.67	938.24	1086.97	0.8711	1303.54	0.4111
1	0.9933	59.06	879.93	1025.94	0.8261	1264.70	0.3957
2	0.9837	57.49	803.98	953.36	0.7694	1223.48	0.3848
3	0.9723	55.63	741.23	897.88	0.7284	1180.15	0.3810
4	0.9592	53.34	680.50	848.30	0.6981	1135.07	0.3808
5	0.9449	50.59	617.69	799.50	0.6495	1088.18	0.3817
6	0.9292	47.03	543.99	743.43	0.6050	1039.03	0.3809
7	0.9127	42.49	464.30	687.39	0.5601	987.06	0.3806
8	0.8954	36.56	373.90	627.67	0.5126	931.58	0.3779
9	0.8769	27.98	265.71	566.36	0.4634	871.30	0.3737
10	0.8579	11.99	106.31	511.80	0.4195	804.63	0.3715
hub	0.8024						
streamline no.	pct. pass.	press. ratio	temp. ratio	local blade forces radius for axial tang. (in.) (lbs/in)	dev. angle (deg)	out. stream. angle (deg)	shock element loss solidify
1	1.69	1.5100	1.1374	9.3705	14.4687	-8.4990	0.306
2	9.14	1.5099	1.1337	9.0809	13.6194	-8.2358	3.43
3	17.04	1.5384	1.1369	8.7755	13.2667	-8.2137	53.93
4	25.34	1.5495	1.1370	8.4532	12.7130	-8.0875	3.72

TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

5	33.98	1.5509	1.1357	8.1171	11.9473	-7.9722	3.91	49.43	
6	42.97	1.5458	1.1340	7.7674	11.0785	-7.8593	4.35	46.24	
7	52.39	1.5490	1.1343	7.4003	10.2062	-7.7991	4.88	42.15	
8	62.35	1.5545	1.1352	7.0095	9.3319	-7.7048	5.62	36.87	
9	72.98	1.5649	1.1367	6.5875	8.3685	-7.5292	6.65	29.91	
10	84.53	1.5864	1.1410	6.1186	7.3275	-7.3006	8.51	19.47	
11	97.31	1.6515	1.1540	5.5723	6.1037	-7.9941	12.86	-0.87	

** values of parameters on streamlines at station, 16, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. angle (deg)	abs.flow slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.f.)	static press. (psia)	static temp. (deg.f.)
tip	9.315	11.010	566.59	367.43	675.42	0.5422	32.96	-1.31	0.010	36.434	695.88	29.950	660.23
1	9.257	11.010	550.48	386.69	672.80	0.5426	35.08	-1.04	0.012	36.356	689.26	29.875	653.83
2	8.989	11.010	539.61	421.10	684.50	0.5534	37.97	-0.71	0.012	36.502	687.28	29.765	650.58
3	8.706	11.010	538.76	439.99	695.60	0.5638	39.24	-0.32	0.010	36.603	684.88	29.621	646.96
4	8.411	11.010	536.59	454.73	703.35	0.5715	40.28	0.15	0.006	36.593	682.37	29.444	643.58
5	8.105	11.010	536.59	469.68	715.95	0.5827	41.00	0.73	0.001	36.629	681.54	29.227	641.33
6	7.786	11.010	540.32	493.02	728.42	0.5937	42.60	1.42	-0.005	36.594	681.10	28.956	639.47
7	7.452	11.010	536.05	519.22	747.44	0.6100	44.00	2.22	-0.012	36.616	681.45	28.609	637.61
8	7.099	11.010	537.26	537.66	767.13	0.6273	46.03	3.18	-0.021	36.542	681.19	28.161	635.00
9	6.721	11.010	531.77	552.59	797.23	0.6530	48.52	4.34	-0.031	36.526	682.71	27.563	632.81
10	6.311	11.010	526.58	582.09	833.42	0.7081	52.53	5.74	-0.047	37.096	689.69	26.698	631.20
11	5.857	11.010	522.61	615.24	863.42								
hub	5.763	11.010											

** values of parameters on streamlines at station, 17, which is the inlet of stator number, 1, of stage number, 2 **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. angle (deg)	abs.flow slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.f.)	static press. (psia)	static temp. (deg.f.)
tip	9.297	11.859	579.48	368.11	686.54	0.5518	32.42	-0.62	0.025	36.435	695.39	29.746	658.55
1	9.240	11.860	562.57	387.24	682.99	0.5513	34.54	-0.59	0.012	36.358	689.19	29.692	652.68
2	8.976	11.864	552.31	421.49	694.78	0.5621	37.35	-0.40	0.005	36.503	687.26	29.577	649.45
3	8.698	11.871	552.52	440.13	706.39	0.5731	38.54	-0.10	0.002	36.603	684.90	29.420	645.79
4	8.409	11.873	551.43	454.52	714.61	0.5822	39.50	0.27	0.000	36.594	682.44	29.233	642.39
5	8.109	11.877	555.91	469.03	727.25	0.5923	40.15	0.67	-0.006	36.630	681.61	29.010	640.11
6	7.797	11.877	555.94	491.72	739.62	0.6034	41.67	1.11	-0.011	36.594	681.18	28.741	638.25
7	7.472	11.882	552.40	517.06	757.52	0.6191	43.02	1.61	-0.011	36.617	681.53	28.405	636.44
8	7.128	11.887	553.94	546.74	776.31	0.6353	44.98	2.18	-0.017	36.543	681.27	27.980	633.96
9	6.763	11.893	548.73	574.13	804.38	0.6593	47.39	2.82	-0.026	36.526	682.75	27.419	631.95
10	6.367	11.904	543.94	591.98	833.42								
11	5.922	11.929	540.01	615.24	863.42								
hub	5.842	11.934											

streamline no. r/rtip	flow coef.	rel.flow angle (deg)
tip	1.0000	
1	0.9939	0.4357
2	0.9655	0.4230
3	0.9356	0.4153
4	0.9044	0.4155
5	0.8722	0.4147

TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

streamline no.	pct. pass.	local blade forces		--- out. stream. ---	
		radius (in.)	tang. (lbs/in)	dev. out. blade angle (deg)	angle (deg)
1	1.64	9.2436	1.8168	8.2495	11.17
2	9.72	8.9747	2.1263	8.1033	10.45
3	17.94	8.6964	2.5607	8.2336	-12.76
4	26.31	8.4100	2.7518	8.2094	10.12
5	34.92	8.1140	2.8679	8.1327	9.83
6	43.87	7.8070	2.9531	8.0424	9.67
7	53.22	7.4858	3.0974	8.0104	9.57
8	63.08	7.1470	3.2468	7.9707	9.61
9	73.57	6.7864	3.4273	7.8970	9.70
10	84.97	6.3953	3.6452	7.9247	9.88
11	97.44	5.9668	4.0972	8.9249	10.56
1					12.72
1					-16.00

** values of parameters on streamlines at station, 19, which is an annulus **													
streamline no.	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static temp. (deg.r.)
tip	9.303	14.440	532.50	532.50	-40.87	534.06	0.4256	-4.39	-0.10	-0.004	35.858	692.92	31.745
1	9.248	14.440	531.03	531.03	-21.32	531.46	0.4247	-2.30	0.20	-0.002	35.839	688.73	31.742
2	8.977	14.440	538.74	538.74	-6.36	538.81	0.4313	-0.74	0.45	0.002	35.975	687.08	31.744
3	8.701	14.440	542.48	542.48	-0.47	542.52	0.4350	-0.05	0.67	0.005	36.064	685.00	31.754
4	8.420	14.440	539.67	539.67	3.54	539.74	0.4334	0.27	0.88	0.009	36.070	682.82	31.771
5	8.131	14.440	532.02	532.02	3.54	532.18	0.4275	0.38	1.09	0.009	36.081	682.05	31.792
6	7.831	14.440	527.71	527.71	3.62	527.89	0.4238	0.51	1.28	0.009	35.981	681.63	31.816
7	7.517	14.440	515.43	515.43	2.45	515.63	0.4138	0.39	1.45	0.009	35.933	681.98	31.840
8	7.186	14.440	501.44	501.44	-5.39	501.47	0.4017	-0.62	1.72	0.006	35.755	681.72	31.860
9	6.833	14.440	500.57	500.57	-28.21	501.63	0.4004	-3.22	1.85	-0.012	35.482	683.02	31.869
10	6.450	14.440										688.12	31.848
11	6.033	14.440											
hub	5.947	14.440											

** values of parameters on streamlines at station, 20, which is an annulus **

streamline no.	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static temp. (deg.r.)
tip	9.298	15.700	558.96	558.96	-40.90	560.47	0.4475	-4.18	-0.40	0.006	35.858	692.59	31.346
1	9.243	15.705	556.35	556.35	-21.31	556.77	0.4457	-2.19	0.39	0.017	35.839	688.64	31.362
2	8.982	15.729	561.17	561.17	-6.95	561.32	0.4499	-0.71	1.09	0.024	35.974	687.03	31.399
3	8.716	15.753	561.45	561.45	-0.47	561.70	0.4509	-0.05	1.69	0.029	36.063	685.00	31.458
4	8.445	15.778	554.93	554.93	2.58	555.35	0.4463	0.27	2.21	0.032	36.049	682.86	31.531
5	8.165	15.803	550.69	550.69	3.53	550.70	0.4427	0.37	2.65	0.036	36.089	682.13	31.613
6	7.873	15.830	538.46	538.46	4.73	539.23	0.4333	0.50	3.00	0.048	36.980	681.71	31.710
7	7.566	15.858											
8	7.244	15.887											

TABLE 2 - EXAMPLE OUTPUT DATA SET (no statior reset)

9	6.892	15.919	510.52	511.41	2.43	511.42	0.4103	0.27	3.38	0.042	35.756	681.80	31.922	661.33
10	6.508	15.954	488.67	489.48	-5.34	489.51	0.3918	-0.62	3.31	0.043	35.535	683.06	32.038	664.32
11	6.084	15.992	477.54	478.14	-27.97	478.96	0.3819	-3.35	2.88	0.045	35.486	687.89	32.160	669.97
hub	6.000	16.000												

1

** values of parameters on streamlines at station, 21, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. angle (deg)	abs. flow stream. slope (deg)	stream. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)	
tip	9.319	17.000	626.06	627.58	-40.80	628.91	0.5045	-3.72	3.99	0.109	35.859	692.24	30.246	661.33
1	9.266	17.010	609.03	611.31	-21.20	611.67	0.4914	-1.99	4.94	0.100	35.840	688.55	30.488	659.28
2	9.029	17.055	600.93	603.98	-6.89	604.02	0.4855	-0.45	5.76	0.094	35.974	686.98	30.717	658.44
3	8.785	17.102	590.29	594.07	-0.47	594.07	0.4779	-0.04	6.47	0.089	36.061	685.01	30.941	657.35
4	8.532	17.150	574.52	578.96	2.54	578.97	0.4660	0.25	7.10	0.085	36.048	682.95	31.159	656.70
5	8.269	17.200	561.25	566.32	3.47	566.33	0.4557	0.35	7.67	0.083	36.057	682.20	31.371	657.09
6	7.993	17.253	541.86	547.45	4.65	547.47	0.4401	0.49	8.20	0.082	35.979	681.73	31.584	658.32
7	7.700	17.309	524.67	530.76	3.52	530.77	0.4262	0.38	8.60	0.083	35.931	681.83	31.796	660.06
8	7.387	17.368	498.56	504.97	2.38	504.97	0.4093	0.27	9.14	0.085	35.757	681.88	32.016	661.92
9	7.049	17.433	468.04	474.66	-5.21	474.69	0.3797	-0.63	9.58	0.090	35.538	683.11	32.243	665.48
10	6.673	17.504	446.26	453.23	-27.23	454.05	0.3616	-3.44	10.06	0.102	35.490	687.68	32.489	671.58
hub	6.171	17.600												

** values of parameters on streamlines at station, 22, which is an annulus **

streamline no.	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. angle (deg)	abs. flow stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	9.406	17.750	639.16	648.41	-40.40	649.67	0.5220	-3.57	9.69	0.135	35.859	692.05	29.892	659.05
1	9.355	17.764	617.82	627.74	-20.96	628.09	0.5052	-1.91	10.20	0.122	35.840	688.49	30.215	657.63
2	9.131	17.828	605.89	616.67	-6.80	616.71	0.4962	-0.63	10.73	0.113	35.973	686.95	30.507	657.19
3	8.899	17.894	591.99	603.61	-0.46	603.61	0.4859	-0.04	11.26	0.107	36.060	685.01	30.784	656.49
4	8.658	17.962	573.39	585.77	2.50	585.77	0.4717	0.24	11.80	0.102	36.047	682.98	31.049	656.12
5	8.407	18.033	557.48	570.69	3.41	570.70	0.4593	0.34	12.35	0.099	36.066	682.25	31.302	656.75
6	8.143	18.108	535.70	549.60	4.55	549.62	0.4419	0.47	12.92	0.096	35.978	681.83	31.551	658.18
7	7.863	18.187	516.34	531.03	3.44	531.04	0.4264	0.37	13.51	0.095	35.931	682.18	31.792	660.11
8	7.564	18.272	488.53	503.80	2.31	503.81	0.4039	0.26	14.15	0.094	35.758	681.92	32.033	662.06
9	7.240	18.364	457.14	472.96	-5.05	472.99	0.3783	-0.61	14.86	0.093	35.541	683.13	32.267	665.64
10	6.881	18.465	436.34	453.25	-26.26	454.01	0.3616	-3.32	15.70	0.088	35.493	687.56	32.492	671.46
hub	6.406	18.600												

1

TABLE 2 - EXAMPLE OUTPUT DATA SET (no stator reset)

6	9.071	20.676	492.75	548.86	3.06	546.87	0.4411	0.32	26.13	0.040	36.063	682.38	31.540	658.80
7	8.844	20.819	479.86	537.20	4.05	537.22	0.4316	0.43	26.71	0.039	35.376	681.97	31.738	659.38
8	8.508	20.967	469.15	529.22	3.02	529.23	0.4248	0.33	27.56	0.038	35.329	682.32	31.818	660.40
9	8.360	21.122	449.79	512.96	2.00	512.96	0.4115	0.22	28.73	0.039	35.760	682.07	31.905	661.47
10	8.096	21.288	426.16	493.78	-4.29	493.80	0.3953	-0.50	30.34	0.040	35.548	683.21	31.991	664.13
11	7.813	21.465	409.35	485.41	-21.78	485.90	0.3877	-2.57	32.51	0.041	35.502	687.17	32.077	668.72
hub	7.758	21.500												

1

TABLE 3 - EXAMPLE OUTPUT DATA SET(with stator resets)

1

*** Input data for axial-flow compressor off-design performance prediction program ***

low aspect ratio two stage nasa fan-- with stator resets

stator reset

fraction design speed	reset guide vane angle	reset stator 1 angle	reset stator 2 angle	reset stator 3 angle	reset stator 4 angle
1.00000	0.00000	-10.00000	5.00000	0.00000	0.00000
0.70000	0.00000	-10.00000	5.00000	0.00000	0.00000

the comp. design rotational speed is 16042.8 rpm.
the design inlet flow rate is 73.300 (lb/sec).
the compressor design pressure ratio is 2.400 .
the range to be covered to the choke side is 0.143 .
the range to be covered to the stall side is 0.190 .

calcs. will be made on 11 streamlines.
the compressor has 4 blade rows.
the molecular weight is 28.96 .
the choke point interval is 0.025 .
the stall point interval is 0.027 .

calculations will be made at the blade edges and at 17 annular stations.

the fractions of design speed to be investigated are 1.000 0.700

the specific heat polynomial is in the following form

$$cp = 0.23762E+00 + 0.39557E-04t + -0.28463E-06t^{**2} + 0.81651E-09t^{**3} + -0.81994E-12t^{**4} + 0.28443E-15t^{**5}$$

input distributions by streamline or streamtube

streamline no.	inlet total temperature (deg. r.)	inlet total pressure (psia)	inlet whirl velocity (ft/sec)	streamtube no.	streamtube flow fraction
1	518.700	14.125	0.000	1	0.1000
2	518.700	14.670	0.000	2	0.2000
3	518.700	14.700	0.000	3	0.3000
4	518.700	14.700	0.000	4	0.4000
5	518.700	14.700	0.000	5	0.5000
6	518.700	14.700	0.000	6	0.6000
7	518.700	14.700	0.000	7	0.7000
8	518.700	14.700	0.000	8	0.8000
9	518.700	14.700	0.000	9	0.9000
10	518.700	14.700	0.000	10	1.0000

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

pct. pass.	d-factor	loss param.	d-factor	loss param.	d-factor	loss param.	d-factor	loss param.	d-factor	loss param.
0.00	0.3000	0.0309	0.4000	0.0336	0.5000	0.0373	0.6000	0.0430	0.7000	0.0508
10.00	0.3000	0.0272	0.4000	0.0290	0.5000	0.0320	0.6000	0.0362	0.7000	0.0423
20.00	0.3000	0.0250	0.4000	0.0263	0.5000	0.0282	0.6000	0.0313	0.7000	0.0360
30.00	0.3000	0.0230	0.4000	0.0239	0.5000	0.0250	0.6000	0.0280	0.7000	0.0310
40.00	0.3000	0.0211	0.4000	0.0220	0.5000	0.0234	0.6000	0.0261	0.7000	0.0286
50.00	0.3000	0.0212	0.4000	0.0222	0.5000	0.0236	0.6000	0.0264	0.7000	0.0299
60.00	0.3000	0.0214	0.4000	0.0226	0.5000	0.0241	0.6000	0.0269	0.7000	0.0306
70.00	0.3000	0.0218	0.4000	0.0231	0.5000	0.0248	0.6000	0.0278	0.7000	0.0317
80.00	0.3000	0.0233	0.4000	0.0248	0.5000	0.0270	0.6000	0.0303	0.7000	0.0347
90.00	0.3000	0.0272	0.4000	0.0290	0.5000	0.0320	0.6000	0.0352	0.7000	0.0423
100.00	0.3000	0.0294	0.4000	0.0317	0.5000	0.0358	0.6000	0.0441	0.7000	0.0486
*** input station data ***										

** input set no. 1 is an annular station **

tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)
-11.0000	-11.0000	0.0000	0.0000

** input set no. 2 is an annular station **

tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)
-9.0000	-9.0000	0.0044	0.0122

** input set no. 3 is an annular station **

tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)
-7.0000	-7.0000	0.0088	0.0249

** input set no. 4 is an annular station **

tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)
-5.2000	-5.2000	0.0133	0.0382

** input set no. 5 is an annular station **

tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)
-3.7000	-3.7000	0.0222	0.0632

** input set no. 6 is an annular station **

tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

1	(inches)	(inches)	(inches)	(inches)					
	-2.3000	-2.6000	0.0289	0.0818					
	*** input station data ***								
	** input set no. 7 is an annular station **								
	tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)					
	-1.0000	-1.5000	0.0354	0.0990					
	*** input station data ***								
	** input set no. 8 is rotor no. 1 **								
	tip c.g. axial location (inches)	hub c.g. axial location (inches)	inlet tip b. layer thick. (inches)	inlet hub b. layer thick. (inches)	blade dihedral angle (degrees)	range improvement bleed option			
	0.9410	0.9410	0.0457	0.1226	0.00			0.00000	
+									
	number of blades	adia. eff. of cum. comp. stages at design	outlet tip b. layer thick. (inches)	outlet hub b. layer thick. (inches)	cum. energy addition fract. thru this stage				
	22	0.8569	0.0524	0.1026	0.5000				

* table of input blade element definition parameters *

strm. no.	l.e. rad. /chord	t.e. rad. /chord	max th. pt. loc. /chord	max th. pt. loc. /chord	trans. pt. location /chord	segment in/out turn rate	in blade angle (deg)	out blade angle (deg)	aero. chord (in.)	layout cone ang. (deg)	stack pt. radius (in.)	l.e. ctr. cen. loc. from stack. line z (in.)	o (deg)
1	0.00180	0.00180	0.02901	0.64000	0.70000	0.07500	67.264	53.820	3.66433	-10.148	9.8555	0.74283	-10.113
2	0.00185	0.00187	0.03000	0.63000	0.64740	0.18000	63.415	52.955	3.67082	-7.508	9.4199	0.87783	-10.289
3	0.00198	0.00203	0.03241	0.62000	0.60420	0.43000	61.346	52.733	3.66862	-5.698	8.9938	0.94321	-10.565
4	0.00219	0.00227	0.03617	0.61000	0.56270	0.66000	59.473	50.579	3.66661	-3.909	8.5539	1.00447	-10.879
5	0.00246	0.00258	0.04115	0.60000	0.51930	0.79000	57.362	47.352	3.66512	-2.152	8.0940	1.07705	-11.212
6	0.00279	0.00296	0.04726	0.58000	0.47050	0.83000	55.066	42.201	3.66436	-0.384	7.6089	1.15829	-11.490
7	0.00319	0.00338	0.05432	0.56000	0.41800	0.86000	52.455	36.314	3.66481	1.443	7.0903	1.25020	-11.752
8	0.00362	0.00382	0.06208	0.54000	0.35920	0.96000	49.456	27.960	3.66763	3.460	6.5284	1.36213	-11.896
9	0.00409	0.00424	0.07007	0.50000	0.28620	0.98000	45.822	17.331	3.67588	5.871	5.9069	1.48369	-11.707
10	0.00453	0.00460	0.07717	0.50000	0.22430	1.00000	41.866	2.559	3.69775	8.994	5.1979	1.64886	-11.117
11	0.00480	0.00480	0.08022	0.50000	0.16290	1.00000	36.961	-16.396	3.76167	13.853	4.3319	1.80300	-9.081
1	* parameters for greater specification of blade element geometry *												
	poly. coef. for 1st seg. centerline angle				poly. coef. for 2nd seg. centerline angle				(function of path dist. from trans. pt.)				
strm. no.	*****				*****				*****				
	linear	quadratic	cubic	quartic	linear	quadratic	cubic	quartic					
1	0.35068	0.69148	0.00000	-0.69148	-0.69642	0.00000	-0.69148	0.00000					
2	0.34861	0.59478	0.00000	-0.59478	-0.64600	0.00000	-0.59478	0.00000					
3	0.34878	0.51486	0.00000	-0.51486	-0.60621	0.00000	-0.51486	0.00000					

TABLE 3 - EXAMPLE OUTPUT DATA SET(with stator resets)

tip c.g. axial location (inches)	hub c.g. axial location (inches)	inlet tip b. layer thick. (inches)	inlet hub b. layer thick. (inches)	blade dihedral angle (degrees)	range improvement bleed option
9.2000	9.2000	0.0663	0.1231	0.00	0.00000

+

number of blades	adia. eff. of cum. comp. stages at design	outlet tip b. layer thick. (inches)	outlet hub b. layer thick. (inches)	cum. energy addition fract. thru this stage
38	0.8526	0.0608	0.0995	1.0000

1.0000

* table of input blade element definition parameters *

strm. no.	l.e.rad. /chord	t.e.rad. /chord	max.th. /chord	max.th. pt.loc. /chord	tran.pt. in/out angle	segment in/out angle	in.blade out.blade angle	aero. chord (in.)	layout cone ang. (deg)	stack pt. radius (in.)	l.e.cir.cen.loc. from stack. line z (in.)	o (deg)
1	0.00614	0.00615	0.03403	0.50000	0.66270	0.61000	62.046	56.549	2.02543	-6.611	9.3752	0.48473
2	0.00682	0.00686	0.03507	0.50000	0.63508	0.66300	61.065	55.480	2.02737	-5.160	9.0680	-5.280
3	0.00751	0.00759	0.03744	0.50000	0.60532	0.76700	59.945	53.732	2.02656	-3.787	8.7545	-5.406
4	0.00822	0.00834	0.04100	0.50000	0.57342	0.88600	58.660	51.711	2.02594	-2.439	8.4319	-5.518
5	0.00895	0.00911	0.04561	0.50000	0.53954	0.98100	57.239	49.221	2.02554	-1.076	8.0986	-5.630
6	0.00972	0.00991	0.05110	0.50000	0.50394	1.00000	55.718	46.031	2.02544	0.303	7.7514	-5.741
7	0.01053	0.01075	0.05726	0.50000	0.46654	1.00000	54.155	41.968	2.02578	1.746	7.3862	-5.850
8	0.01140	0.01163	0.06383	0.50000	0.42725	1.00000	52.589	36.760	2.02689	3.363	6.9982	-5.952
9	0.01235	0.01256	0.07038	0.50000	0.40000	1.00000	51.128	30.009	2.02957	5.318	6.5811	-6.043
10	0.01341	0.01357	0.07622	0.50000	0.40000	1.00000	50.352	20.164	2.03577	7.811	6.1192	-6.121
11	0.01474	0.01476	0.07989	0.50000	0.40000	1.00000	52.662	-0.228	2.05176	11.213	5.5584	-6.180

*** input station data ***

** input set no. 13 is an annular station **

tip axial location (inches)	hub axial location (inches)	tip b. layer thick. (inches)	hub b. layer thick. (inches)
11.0100	11.0100	0.0577	0.0928

1

*** input station data ***

** input set no. 14 is a guide vane or stator **

tip c.g. axial location (inches)	hub c.g. axial location (inches)	inlet tip b. layer thick. (inches)	inlet hub b. layer thick. (inches)	blade dihedral angle (degrees)	range improvement bleed option
12.7000	12.7000	0.0564	0.0890	0.00	0.00000

+

number of blades	blade rotation option for stall margin	outlet tip b. layer thick. (inches)	outlet hub b. layer thick. (inches)
42	0.0556	0.0866	0.0866

42

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

* table of input blade element definition parameters *													
strm. no.	l.e. rad. /chord	t.e. rad. /chord	max. th. /chord	max. th. pt. loc. /chord	tran. pt. location	segment in/out	in-blade angle (deg)	out-blade angle (deg)	aero. chord (in.)	layout cone ang. (deg)	stack pt. radius (in.)	l.e. cir. cen. loc. from stack. line z (in.)	o (deg)
1	0.01387	0.01387	0.07967	0.50000	0.40000	1.00000	35.156	-15.591	1.75296	0.218	9.2437	0.81577	1.758
2	0.01323	0.01323	0.07807	0.50000	0.40000	1.00000	34.571	-12.796	1.75301	0.413	8.9689	0.81297	1.857
3	0.01257	0.01257	0.07643	0.50000	0.40000	1.00000	34.651	-10.900	1.75305	0.584	8.6891	0.80990	1.981
4	0.01190	0.01190	0.07476	0.50000	0.40000	1.00000	34.762	-9.901	1.75310	0.743	8.4021	0.80850	2.091
5	0.01121	0.01121	0.07304	0.50000	0.40000	1.00000	35.247	-9.401	1.75316	0.900	8.1069	0.80689	2.222
6	0.01050	0.01049	0.07123	0.50000	0.40000	1.00000	36.187	-9.202	1.75324	1.077	7.8010	0.80427	2.390
7	0.00975	0.00975	0.06938	0.50000	0.40000	1.00000	37.386	-9.103	1.75335	1.280	7.4815	0.80084	2.594
8	0.00897	0.00896	0.06742	0.50000	0.40000	1.00000	38.803	-9.304	1.75349	1.497	7.1458	0.79710	2.831
9	0.00814	0.00813	0.06535	0.50000	0.40000	1.00000	40.462	-9.603	1.75364	1.703	6.7902	0.79249	3.117
10	0.00724	0.00723	0.06311	0.50000	0.40000	1.00000	43.769	-11.099	1.75379	1.906	6.4061	0.78390	3.556
11	0.00621	0.00620	0.06051	0.50000	0.40000	1.00000	52.207	-15.990	1.75422	2.433	5.9636	0.76049	4.456
*** input station data ***													
** input set no. 15 is an annular station **													
tip axial location (inches)		14.4400		hub axial location (inches)		14.4400		tip b. layer thick. (inches)		0.0552		hub b. layer thick. (inches)	
** input set no. 16 is an annular station **													
tip axial location (inches)		15.7000		hub axial location (inches)		16.0000		tip b. layer thick. (inches)		0.0549		hub b. layer thick. (inches)	
** input set no. 17 is an annular station **													
tip axial location (inches)		17.0000		hub axial location (inches)		17.6000		tip b. layer thick. (inches)		0.0538		hub b. layer thick. (inches)	
** input set no. 18 is an annular station **													
tip axial location (inches)		17.7500		hub axial location (inches)		18.6000		tip b. layer thick. (inches)		0.0529		hub b. layer thick. (inches)	
** input set no. 19 is an annular station **													
tip axial location (inches)				hub axial location (inches)				tip b. layer thick. (inches)		0.0765		hub b. layer thick. (inches)	

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

18.5000 19.6000 0.0521 0.0730

tip axial location (inches) hub axial location (inches) tip b. layer thick. (inches) hub b. layer thick. (inches)

** input set no. 20 is an annular station **

19.2500 20.6000 0.0511 0.0687

*** input station data ***

** input set no. 21 is an annular station **

tip axial location (inches) hub axial location (inches) tip b. layer thick. (inches) hub b. layer thick. (inches)

20.0000 21.5000 0.0500 0.0649

radius array

i	1	2	3	4	5	6	7	8	9	10	11
1	10.0990	9.6525	9.1843	8.6909	8.1678	7.6088	7.0053	6.3447	5.6067	4.7556	3.7144
2	10.0948	9.6462	9.1758	8.6799	8.1539	7.5915	6.9840	6.3184	5.5738	4.7130	3.6548
3	10.0912	9.6407	9.1680	8.6697	8.1409	7.5753	6.9639	6.2933	5.5422	4.6720	3.5969
4	10.0878	9.6356	9.1611	8.6606	8.1294	7.5609	6.9461	6.2713	5.5145	4.6358	3.5457
5	10.0847	9.6283	9.1558	8.6556	8.1288	7.5532	6.9317	6.2609	5.5094	4.6282	3.5310
6	10.0818	9.6215	9.1513	8.6513	8.1257	7.5495	6.9280	6.2505	5.4980	4.6154	3.5180
7	10.0792	9.6149	9.1473	8.6477	8.1227	7.5453	6.9248	6.2463	5.4945	4.6115	3.5140
8	10.0767	9.6116	9.1446	8.6450	8.1200	7.5426	6.9221	6.2436	5.4918	4.6088	3.5112
9	10.0743	9.6084	9.1414	8.6422	8.1172	7.5398	6.9193	6.2408	5.4890	4.6060	3.5084
10	10.0720	9.6052	9.1382	8.6394	8.1144	7.5370	6.9165	6.2380	5.4862	4.6032	3.5056
11	10.0697	9.6020	9.1350	8.6366	8.1116	7.5342	6.9137	6.2352	5.4834	4.6004	3.5028
12	10.0674	9.5988	9.1318	8.6338	8.1088	7.5314	6.9109	6.2324	5.4806	4.5976	3.4999
13	10.0651	9.5956	9.1286	8.6310	8.1060	7.5286	6.9081	6.2296	5.4778	4.5948	3.4971
14	10.0628	9.5924	9.1254	8.6282	8.1032	7.5258	6.9053	6.2268	5.4750	4.5920	3.4943
15	10.0605	9.5892	9.1222	8.6254	8.1004	7.5230	6.9025	6.2240	5.4722	4.5892	3.4915
16	10.0582	9.5860	9.1190	8.6226	8.0976	7.5202	6.8997	6.2212	5.4694	4.5864	3.4887
17	10.0559	9.5828	9.1158	8.6198	8.0948	7.5174	6.8969	6.2184	5.4666	4.5836	3.4859
18	10.0536	9.5796	9.1126	8.6170	8.0920	7.5146	6.8941	6.2156	5.4638	4.5808	3.4831
19	10.0513	9.5764	9.1094	8.6142	8.0892	7.5118	6.8913	6.2128	5.4610	4.5780	3.4803
20	10.0490	9.5732	9.1062	8.6114	8.0864	7.5090	6.8885	6.2100	5.4582	4.5752	3.4775
21	10.0467	9.5700	9.1030	8.6086	8.0836	7.5062	6.8857	6.2072	5.4554	4.5724	3.4747
22	10.0444	9.5668	9.1002	8.6058	8.0808	7.5034	6.8829	6.2044	5.4526	4.5696	3.4719
23	10.0421	9.5636	9.0974	8.6030	8.0780	7.5006	6.8801	6.2016	5.4498	4.5668	3.4691
24	10.0398	9.5604	9.0946	8.6002	8.0752	7.4978	6.8773	6.1988	5.4470	4.5640	3.4663
25	10.0375	9.5572	9.0918	8.5974	8.0724	7.4950	6.8745	6.1960	5.4442	4.5612	3.4635

z (ft, fm) i z (i, j) ar

1 -11.0000 1 -11.0000 0.0000

2 -9.0000 2 -9.0000 3.2200

3 -7.0000 3 -7.0000 3.2471

4 -5.2000 4 -5.2000 3.6345

5 -3.7000 5 -3.7000 4.3318

6 -2.4158 6 -2.4158 5.0202

TABLE 3 - EXAMPLE OUTPUT DATA SET(with stator resets)

z array												
i	*	1	2	3	4	5	6	7	8	9	10	11
1	*	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000	-11.0000
2	*	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000	-9.0000
3	*	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000	-7.0000
4	*	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000	-5.2000
5	*	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000	-3.7000
6	*	-2.3013	-2.3218	-2.3433	-2.3660	-2.3901	-2.4158	-2.4436	-2.4741	-2.5081	-2.5476	-2.5962
7	*	-1.0027	-1.0371	-1.0731	-1.1110	-1.1512	-1.1941	-1.2405	-1.2911	-1.3477	-1.4128	-1.4821
8	*	-0.9429	-0.7890	-0.6785	-0.6087	-0.5633	-0.5437	-0.5600	-0.6158	-0.7618	-0.9417	-1.1249
9	*	1.5889	2.0072	2.2943	2.4444	2.5573	2.6524	2.7209	2.7631	2.7866	2.7928	2.7828
10	*	3.0034	3.0257	3.0489	3.0731	3.0985	3.1253	3.1537	3.1841	3.2169	3.2529	3.2932
11	*	4.1826	4.1922	4.1947	4.1959	4.1964	4.1961	4.1955	4.1945	4.1925	4.1864	4.1742
12	*	6.3435	6.3343	6.3326	6.3325	6.3331	6.3346	6.3367	6.3394	6.3430	6.3492	6.3592
13	*	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400	7.3400
14	*	8.7029	8.6845	8.6610	8.6340	8.6037	8.5705	8.5341	8.4939	8.4507	8.4027	8.3373
15	*	9.7328	9.7515	9.7763	9.8038	9.8351	9.8705	9.9123	9.9596	10.0125	10.0734	10.1427
16	*	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100	11.0100
17	*	11.8599	11.8638	11.8681	11.8707	11.8736	11.8776	11.8825	11.8877	11.8940	11.9050	11.9286
18	*	13.5877	13.5857	13.5840	13.5830	13.5825	13.5824	13.5823	13.5827	13.5832	13.5857	13.5931
19	*	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400	14.4400
20	*	15.7050	15.7292	15.7541	15.7798	15.8064	15.8340	15.8628	15.8928	15.9242	15.9573	15.9924
21	*	17.0102	17.0590	17.1092	17.1610	17.2145	17.2699	17.3274	17.3874	17.4500	17.5157	17.5850
22	*	17.7645	17.8344	17.9062	17.9801	18.0563	18.1351	18.2167	18.3015	18.3898	18.4822	18.5791
23	*	18.5190	18.6106	18.7046	18.8011	18.9004	19.0028	19.1086	19.2181	19.3317	19.4500	19.5735
24	*	19.2737	19.3879	19.5049	19.6247	19.7477	19.8740	20.0040	20.1380	20.2764	20.4196	20.5683
25	*	20.0268	20.1557	20.2873	20.4218	20.5594	20.7003	20.8449	20.9933	21.1459	21.3032	21.4655
to array												
i	*	1	2	3	4	5	6	7	8	9	10	11
1	*	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70
2	*	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70
3	*	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70
4	*	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70
5	*	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70
6	*	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70
7	*	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70	518.70

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

[illegible]

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

		1		2		3		4		5		6		7		8		9		10		11	
		i		j		k		l		m		n		o		p		q		r		s	
1	*	381.89	397.29	414.72	434.67	457.80	485.07	517.88	558.39	610.19	679.72	780.15											
2	*	386.38	401.57	418.71	438.25	460.80	487.24	518.82	557.47	606.28	670.65	761.15											
3	*	387.88	402.83	419.67	438.80	460.82	486.52	517.06	554.19	600.67	661.23	744.79											
4	*	19.45	20.18	20.99	21.91	22.96	24.17	25.61	27.33	29.46	32.17	35.81											
5	*	19.45	20.15	20.95	21.85	22.88	24.07	25.47	27.14	29.19	31.79	35.24											
6	*	19.65	20.34	21.12	21.99	22.98	24.12	25.45	27.02	28.93	31.30	34.37											
7	*	417.13	430.68	445.65	462.29	480.94	502.05	526.21	554.23	587.26	637.01	676.09											
8	*	419.66	432.84	447.34	463.41	481.34	501.54	524.50	550.95	581.94	618.59	663.31											
9	*	420.40	433.34	447.54	463.25	480.73	500.35	522.58	548.07	577.70	612.72	654.99											
10	*	18.16	18.71	19.32	19.98	20.72	21.55	22.48	23.55	24.79	26.24	27.99											
11	*	18.16	18.71	19.31	19.96	20.70	21.51	22.44	23.49	24.71	26.14	27.84											
12	*	18.17	18.71	19.30	19.95	20.67	21.47	22.37	23.40	24.58	25.96	27.61											
13	*	18.13	18.64	19.20	19.82	20.50	21.25	22.10	23.05	24.14	25.40	26.88											
14	*	17.95	18.44	18.97	19.54	20.18	20.87	21.65	22.52	23.50	24.62	25.92											
15	*	17.64	18.09	18.57	19.09	19.66	20.29	20.98	21.74	22.59	23.56	24.65											
16	*	17.20	17.59	18.02	18.47	18.97	19.50	20.08	20.72	21.42	22.20	23.07											
17	*	16.97	17.33	17.72	18.14	18.60	19.10	19.61	20.18	20.81	21.50												
18	*	16.62	16.97	17.33	17.72	18.14	18.60	19.10	19.61	20.18	20.81												
19	*	16.62	16.97	17.33	17.72	18.14	18.60	19.10	19.61	20.18	20.81												
20	*	16.62	16.97	17.33	17.72	18.14	18.60	19.10	19.61	20.18	20.81												
21	*	16.62	16.97	17.33	17.72	18.14	18.60	19.10	19.61	20.18	20.81												
22	*	16.62	16.97	17.33	17.72	18.14	18.60	19.10	19.61	20.18	20.81												
23	*	16.62	16.97	17.33	17.72	18.14	18.60	19.10	19.61	20.18	20.81												
24	*	16.62	16.97	17.33	17.72	18.14	18.60	19.10	19.61	20.18	20.81												
25	*	16.62	16.97	17.33	17.72	18.14	18.60	19.10	19.61	20.18	20.81												
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TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

i	1	2	3	4	5	6	7	8	9	10	11
10 *	-0.0437	-0.0594	-0.0623	-0.0536	-0.0393	-0.0197	0.0053	0.0357	0.0667	0.1052	0.1863
11 *	-0.0169	-0.0234	-0.0245	-0.0196	-0.0113	0.0004	0.0161	0.0357	0.0569	0.0933	0.1271
12 *	0.0143	0.0217	0.0286	0.0347	0.0407	0.0465	0.0523	0.0581	0.0647	0.0721	0.0786
13 *	-0.0202	-0.0184	-0.0156	-0.0113	-0.0055	0.0023	0.0125	0.0259	0.0433	0.0668	0.1013
14 *	-0.1177	-0.0911	-0.0646	-0.0385	-0.0129	0.0127	0.0387	0.0658	0.0954	0.1293	0.1711
15 *	-0.0882	-0.0639	-0.0396	-0.0160	0.0071	0.0298	0.0525	0.0756	0.1005	0.1289	0.1632
16 *	-0.0228	-0.0183	-0.0132	-0.0071	0.0003	0.0092	0.0200	0.0334	0.0502	0.0718	0.1005
17 *	-0.0109	-0.0042	0.0028	0.0099	0.0172	0.0246	0.0323	0.0403	0.0481	0.0558	0.0620
18 *	0.0054	0.0071	0.0089	0.0109	0.0131	0.0156	0.0184	0.0218	0.0257	0.0304	0.0368
19 *	-0.0017	-0.0008	0.0035	0.0062	0.0091	0.0122	0.0155	0.0191	0.0230	0.0273	0.0323
20 *	-0.0069	-0.0022	0.0026	0.0075	0.0126	0.0180	0.0236	0.0293	0.0358	0.0427	0.0503
21 *	0.0697	0.0783	0.0872	0.0963	0.1058	0.1157	0.1262	0.1374	0.1495	0.1627	0.1774
22 *	0.1708	0.1792	0.1879	0.1969	0.2064	0.2165	0.2272	0.2388	0.2514	0.2654	0.2812
23 *	0.2687	0.2783	0.2883	0.2987	0.3098	0.3216	0.3342	0.3478	0.3628	0.3794	0.3980
24 *	0.3879	0.3991	0.4108	0.4231	0.4362	0.4501	0.4650	0.4812	0.4990	0.5185	0.5403
25 *	0.5024	0.5110	0.5203	0.5304	0.5415	0.5537	0.5671	0.5819	0.5983	0.6166	0.6373

xcurve array

i	1	2	3	4	5	6	7	8	9	10	11
1 *	0.0001	0.0001	0.0002	0.0002	0.0003	0.0004	0.0005	0.0006	0.0007	0.0009	0.0010
2 *	0.0002	0.0001	0.0001	0.0001	0.0001	0.0000	0.0000	0.0000	-0.0001	-0.0002	-0.0002
3 *	0.0001	-0.0001	0.0002	0.0006	0.0010	0.0016	0.0022	0.0030	0.0040	0.0055	-0.0066
4 *	-0.0001	-0.0013	-0.0003	0.0007	0.0018	0.0031	0.0046	0.0065	0.0090	0.0127	0.0164
5 *	-0.0049	0.0022	0.0026	-0.0034	0.0044	0.0058	0.0075	0.0099	0.0132	0.0184	-0.0031
6 *	0.0212	-0.0098	-0.0045	-0.0002	0.0039	0.0082	0.0130	0.0187	0.0263	0.0368	0.0762
7 *	-0.1086	-0.0227	0.0158	-0.0106	-0.0057	-0.0006	0.0048	0.0110	0.0184	0.0275	0.0614
8 *	-0.1046	-0.0400	-0.0277	-0.0200	-0.0136	-0.0077	-0.0018	-0.0045	-0.0124	-0.0218	-0.0432
9 *	0.0586	0.0947	0.0937	0.0626	0.0014	0.0001	-0.0014	-0.0033	-0.0060	-0.0096	-0.0417
10 *	0.0344	0.0287	0.0227	0.0168	0.0110	0.0050	-0.0013	-0.0080	-0.0160	-0.0263	-0.0414
11 *	0.0131	0.0238	0.0226	0.0204	0.0173	0.0132	0.0079	0.0012	-0.0064	-0.0158	-0.0392
12 *	0.0162	-0.0076	-0.0069	-0.0069	-0.0071	-0.0072	-0.0073	-0.0072	-0.0059	-0.0034	-0.0036
13 *	-0.0813	-0.0407	-0.0342	-0.0275	-0.0206	-0.0134	-0.0058	0.0026	0.0120	0.0233	0.0467
14 *	-0.0413	-0.0140	-0.0081	-0.0025	0.0028	0.0077	0.0121	0.0160	0.0197	0.0232	0.0260
15 *	0.0936	0.0264	0.0191	0.0120	0.0052	-0.0012	-0.0072	-0.0131	-0.0188	-0.0249	-0.0332
16 *	0.0093	0.0257	0.0193	0.0128	0.0063	0.0067	0.0040	0.0005	-0.0152	-0.0242	-0.0355
17 *	0.0251	0.0119	0.0105	0.0088	0.0067	0.0040	0.0004	-0.0040	-0.0095	-0.0178	-0.0271
18 *	-0.0076	0.0005	-0.0004	-0.0014	-0.0025	-0.0036	-0.0048	-0.0051	-0.0075	-0.0090	-0.0121
19 *	-0.0043	-0.0013	-0.0002	0.0009	0.0019	0.0029	0.0038	0.0045	0.0052	0.0057	0.0059
20 *	0.0060	0.0309	0.0324	0.0340	0.0354	0.0369	0.0384	0.0399	0.0415	0.0431	0.0447
21 *	0.1095	0.0891	0.0881	0.0873	0.0865	0.0858	0.0853	0.0848	0.0845	0.0843	0.0847
22 *	0.1352	0.1197	0.1167	0.1138	0.1111	0.1085	0.1061	0.1038	0.1015	0.0994	0.0984
23 *	0.1114	0.1241	0.1211	0.1182	0.1155	0.1129	0.1104	0.1080	0.1057	0.1033	0.1106
24 *	0.1513	0.1158	0.1117	0.1078	0.1042	0.1007	0.0975	0.0943	0.0912	0.0882	0.0942
25 *	0.0666	0.0836	0.0796	0.0759	0.0725	0.0692	0.0661	0.0631	0.0603	0.0574	0.0415

*** computed compressor parameters for a rotational speed of 16042.8 rpm which is a speed fraction of 1.0000 ***

** the corrected weightflow per unit of casing annular area at the inlet face of the first blade row is 38.90 lbs/sec/ft sq **

stage blade no.	type	flow coef.	head coef.	id. head coef.	press. ratio	temp. ratio	adia. eff.	poly. eff.	reset angle (deg)	for ax. thrust (lbs)	gas bending moments (ft-lbs)	torque (ft-lbs)	power (hp)
1	rotor	0.4446	0.2411	0.2629	1.6460	1.1605	0.9169	0.9223	0.00	1057.92	17.671	-12.757	703.71
1	stator	0.4159	0.2207	0.2629	1.5822	1.1605	0.8395	0.8492	-10.01	-229.94	1.918	7.047	2149.53

** mass averaged rotor and stage aerodynamic parameters **

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

2	rotor	0.4774	0.3133	0.3366	1.6497	1.1556	0.9307	0.9352	0.00	1563.20	8.841	-5.775	807.37	2466.15
2	stator	0.4065	0.3021	0.3366	1.6221	1.1577	0.8973	0.8946	5.00	-273.17	1.439	2.994		

** cumulative sums of mass averaged rotor and stage aerodynamic parameters **

stage blade no. type	weight (lbs/sec)	total press. (psia)	total temp. (deg. F.)	press. ratio	temp. ratio	head coef.	ideal head coef.	adla. eff.	poly. eff.	for. ax. thrust (lbs)	torque (ft-lbs)	power (hp)	fract energy
1	inlet	73.30	14.666	518.70									
1	rotor	73.30	24.140	601.95	1.6460	1.1605	0.2411	0.2629	0.9169	0.9223	1037.92	703.71	2149.53
1	stator	73.30	23.204	601.95	1.5822	1.1605	0.2207	0.2629	0.8395	0.8492	827.97		0.4624
2	rotor	73.30	38.280	696.86	2.6102	1.3435	0.4954	0.5686	0.8712	0.8865	2250.97	1521.84	4648.54
2	stator	73.30	37.641	696.86	2.5666	1.3435	0.4855	0.5686	0.8539	0.8709	1977.79		1.0000

** values of parameters on streamlines at station, 1, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg. F.)	static press. (psia)	static temp. (deg. F.)
tip	10.099	-11.000	514.73	514.73	0.00	514.73	0.4731	0.00	-0.13	0.000	14.144	518.70	497.22
1	10.095	-11.000	571.93	571.93	0.00	571.93	0.5282	0.00	-0.15	0.000	14.670	518.70	492.17
2	9.632	-11.000	574.89	574.89	0.00	574.89	0.5311	0.00	-0.22	0.000	14.700	518.70	491.89
3	9.167	-11.000	574.81	574.81	0.00	574.81	0.5310	0.00	-0.29	0.000	14.700	518.70	491.90
4	8.675	-11.000	574.81	574.81	0.00	574.81	0.5310	0.00	-0.37	0.000	14.700	518.70	491.91
5	8.154	-11.000	574.69	574.69	0.00	574.69	0.5309	0.00	-0.47	0.000	14.700	518.70	491.92
6	7.597	-11.000	574.54	574.54	0.00	574.54	0.5308	0.00	-0.59	0.001	14.700	518.70	491.94
7	6.996	-11.000	574.34	574.34	0.00	574.34	0.5306	0.00	-0.74	0.001	14.700	518.70	491.96
8	6.338	-11.000	574.09	574.09	0.00	574.09	0.5304	0.00	-0.93	0.001	14.700	518.70	491.98
9	5.603	-11.000	573.80	573.80	0.00	573.80	0.5301	0.00	-1.22	0.000	14.700	518.70	492.01
10	4.755	-11.000	573.47	573.47	0.00	573.47	0.5298	0.00	-1.84	0.001	14.660	518.70	492.43
11	3.714	-11.000	568.75	568.75	0.00	568.75	0.5254	0.00					
hub	3.714	-11.000											

** values of parameters on streamlines at station, 2, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg. F.)	static press. (psia)	static temp. (deg. F.)
tip	10.099	-9.000	515.97	515.97	0.00	515.97	0.4742	0.00	-0.11	0.000	14.180	518.70	497.11
1	10.095	-9.000	569.12	569.12	0.00	569.12	0.5255	0.00	-0.13	0.000	14.669	518.70	492.43
2	9.627	-9.000	571.97	571.97	0.00	571.97	0.5283	0.00	-0.18	0.000	14.698	518.70	492.16
3	9.160	-9.000	571.81	571.81	0.00	571.81	0.5281	0.00	-0.24	0.001	14.698	518.70	492.18
4	8.666	-9.000	571.60	571.60	0.00	571.60	0.5279	0.00	-0.31	0.001	14.698	518.70	492.20
5	8.142	-9.000	571.32	571.32	0.00	571.32	0.5276	0.00	-0.39	0.001	14.698	518.70	492.22
6	7.582	-9.000	570.94	570.94	0.00	570.94	0.5273	0.00	-0.50	0.001	14.698	518.70	492.26
7	6.977	-9.000	570.48	570.48	0.00	570.48	0.5269	0.00	-0.65	0.001	14.698	518.70	492.30
8	6.314	-9.000											

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

9	5.571	-9.000	569.95	570.01	0.00	570.01	0.5264	0.00	-0.85	0.001	14.598	518.70	12.188	492.34
10	4.713	-9.000	569.36	569.48	0.00	569.48	0.5258	0.00	-1.15	0.001	14.598	518.70	12.193	492.39
11	3.655	-9.000	564.35	564.55	0.00	564.55	0.5211	0.00	-1.54	0.002	14.558	518.70	12.200	492.85
hub	3.643	-9.000												

1

** values of parameters on streamlines at station, 3, which is an annulus **

streamline no.	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	10.101	-7.000	517.71	517.71	0.00	517.71	0.4759	0.00	-0.09	0.000	14.211	518.70	12.186	496.37
1	10.091	-7.000	567.30	567.30	0.00	567.30	0.5237	0.00	-0.08	0.000	14.568	518.70	12.186	492.59
2	9.624	-7.000	570.04	570.04	0.00	570.04	0.5264	0.00	-0.11	0.001	14.596	518.70	12.187	492.34
3	9.184	-7.000	569.79	569.79	0.00	569.79	0.5261	0.00	-0.14	0.001	14.596	518.70	12.189	492.36
4	8.659	-7.000	569.39	569.40	0.00	569.40	0.5258	0.00	-0.18	0.002	14.596	518.70	12.192	492.40
5	8.133	-7.000	568.80	568.80	0.00	568.80	0.5252	0.00	-0.22	0.002	14.596	518.70	12.192	492.46
6	7.571	-7.000	567.94	567.95	0.00	567.95	0.5244	0.00	-0.29	0.003	14.596	518.70	12.204	492.53
7	6.963	-7.000	566.75	566.76	0.00	566.76	0.5232	0.00	-0.40	0.003	14.596	518.70	12.214	492.64
8	6.295	-7.000	565.17	565.20	0.00	565.20	0.5217	0.00	-0.60	0.004	14.596	518.70	12.226	492.79
9	5.546	-7.000	563.29	563.37	0.00	563.37	0.5199	0.00	-0.97	0.003	14.596	518.70	12.241	492.96
10	4.676	-7.000	559.62	560.00	0.00	560.00	0.5166	0.00	-2.11	-0.007	14.656	518.70	12.236	493.26
11	3.597	-7.000												
hub	3.572	-7.000												

** values of parameters on streamlines at station, 4, which is an annulus **

streamline no.	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	10.101	-5.200	520.23	520.23	0.00	520.23	0.4783	0.00	-0.16	-0.002	14.235	518.70	12.188	496.76
1	10.088	-5.200	567.43	567.43	0.00	567.43	0.5239	0.00	-0.13	-0.001	14.567	518.70	12.184	492.58
2	9.621	-5.200	570.50	570.50	0.00	570.50	0.5268	0.00	-0.09	0.000	14.595	518.70	12.182	492.30
3	9.152	-5.200	570.43	570.43	0.00	570.43	0.5268	0.00	-0.04	0.001	14.595	518.70	12.182	492.31
4	8.656	-5.200	569.95	569.95	0.00	569.95	0.5263	0.00	0.02	0.002	14.595	518.70	12.186	492.35
5	8.131	-5.200	568.99	568.99	0.00	568.99	0.5254	0.00	0.08	0.004	14.595	518.70	12.194	492.44
6	7.568	-5.200	567.40	567.40	0.00	567.40	0.5238	0.00	0.14	0.005	14.595	518.70	12.207	492.59
7	6.959	-5.200	564.88	564.88	0.00	564.88	0.5214	0.00	0.18	0.008	14.595	518.70	12.228	492.82
8	6.290	-5.200	560.91	560.91	0.00	560.91	0.5175	0.00	0.17	0.011	14.595	518.70	12.260	493.18
9	5.537	-5.200	554.14	554.14	0.00	554.14	0.5110	0.00	0.08	0.017	14.695	518.70	12.315	493.79
10	4.658	-5.200	531.82	531.82	0.00	531.82	0.4895	0.00	0.02	0.043	14.655	518.70	12.458	495.76
11	3.546	-5.200												
hub	3.507	-5.200												

1

TABLE 3 - EXAMPLE OUTPUT DATA SET (with statior resets)

** values of parameters on streamlines at station, 5, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip 10.101	-3.700	527.37	527.37	0.00	527.37	0.4852	0.00	-0.65	-0.005	14.254	518.70	12.150	496.15
1	10.079	527.23	527.23	0.00	527.24	0.5295	0.00	-0.29	-0.001	14.666	518.70	12.144	492.14
2	9.615	575.06	575.06	0.00	575.06	0.5313	0.00	-0.10	0.000	14.694	518.70	12.143	491.87
3	9.149	574.83	574.83	0.00	574.83	0.5310	0.00	0.06	0.001	14.694	518.70	12.145	491.90
4	8.656	574.28	574.28	0.00	574.28	0.5305	0.00	0.24	0.002	14.694	518.70	12.150	491.95
5	8.134	573.19	573.19	0.00	573.20	0.5295	0.00	0.44	0.004	14.694	518.70	12.159	492.05
6	7.575	571.16	571.16	0.00	571.20	0.5275	0.00	0.69	0.007	14.694	518.70	12.175	492.23
7	6.970	567.66	567.66	0.00	567.75	0.5242	0.00	0.99	0.011	14.694	518.70	12.204	492.55
8	6.305	562.13	562.13	0.00	562.13	0.5187	0.00	1.32	0.015	14.694	518.70	12.249	493.07
9	5.556	553.27	553.27	0.00	553.50	0.5104	0.00	1.65	0.020	14.694	518.70	12.319	493.85
10	4.681	544.21	544.21	0.00	544.43	0.5016	0.00	1.64	-0.003	14.654	518.70	12.358	494.66
11	3.581												
hub 3.518	-3.700												

** values of parameters on streamlines at station, 6, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip 10.099	-2.300	538.46	538.46	0.00	538.49	0.4959	0.00	0.59	0.021	14.270	518.70	12.080	495.18
1	10.070	577.51	577.51	0.00	577.51	0.5327	0.00	0.09	0.002	14.666	518.70	12.100	491.64
2	9.611	580.60	580.60	0.00	580.60	0.5357	0.00	-0.05	-0.004	14.693	518.70	12.097	491.35
3	9.147	581.90	581.90	0.00	581.90	0.5379	0.00	0.03	-0.005	14.693	518.70	12.086	491.23
4	8.658	582.97	582.97	0.00	582.97	0.5390	0.00	0.28	-0.003	14.693	518.70	12.077	491.13
5	8.140	582.96	582.96	0.00	582.96	0.5390	0.00	0.68	0.001	14.693	518.70	12.077	491.13
6	7.588	581.23	581.23	0.00	581.23	0.5373	0.00	1.20	0.007	14.693	518.70	12.091	491.29
7	6.991	576.88	576.88	0.00	576.88	0.5330	0.00	1.84	0.014	14.693	518.70	12.128	491.70
8	6.335	568.72	568.72	0.00	568.72	0.5251	0.00	2.59	0.022	14.693	518.70	12.195	492.46
9	5.597	554.45	554.45	0.00	554.45	0.5113	0.00	3.40	0.034	14.693	518.70	12.311	493.77
10	4.730	517.75	517.75	0.00	517.75	0.4759	0.00	3.79	0.076	14.653	518.70	12.565	496.96
11	3.623												
hub 3.540	-2.600												

** values of parameters on streamlines at station, 7, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip 10.100	-1.000	503.22	504.06	0.00	504.06	0.4629	0.00	-3.30	-0.109	14.283	518.70	12.348	498.10
1	10.065	567.58	568.25	0.00	568.25	0.5246	0.00	-2.78	-0.067	14.665	518.70	12.176	492.51
2	9.594	585.14	585.57	0.00	585.57	0.5415	0.00	-2.19	-0.044	14.692	518.70	12.055	490.88
3	9.129	595.39	595.59	0.00	595.59	0.5513	0.00	-1.48	-0.029	14.692	518.70	11.970	489.92
4	8.646	601.86	601.89	0.00	601.89	0.5575	0.00	-0.62	-0.017	14.692	518.70	11.916	489.30
5	8.139	-1.152											

TABLE 3 - EXAMPLE OUTPUT DATA SET(with stator resets)

streamline no.	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip 10.034	0.197	0.164	583.64	584.91	0.00	584.91	0.5409	0.00	-3.79	-0.105	14.295	518.70	11.734	490.94
1	9.313	0.049	633.25	637.42	0.00	637.42	0.5925	0.00	-6.56	-0.025	14.665	518.70	11.586	485.72
2	9.450	-0.008	641.63	643.79	0.00	643.79	0.5988	0.00	-4.70	-0.018	14.691	518.70	11.550	485.06
3	9.596	-0.072	645.20	646.15	0.00	646.15	0.6011	0.00	-3.11	-0.012	14.691	518.70	11.529	484.81
4	9.742	-0.146	645.58	645.82	0.00	645.82	0.6008	0.00	-1.58	-0.007	14.691	518.70	11.532	484.84
5	9.889	-0.228	643.01	643.01	0.00	643.01	0.5980	0.00	0.05	-0.002	14.691	518.70	11.557	485.14
6	10.034	-0.322	636.98	637.30	0.00	637.30	0.5923	0.00	1.80	0.004	14.691	518.70	11.608	485.73
7	10.180	-0.435	627.04	628.31	0.00	628.31	0.5835	0.00	3.65	0.010	14.691	518.70	11.687	486.66
8	10.326	-0.561	612.12	615.15	0.00	615.15	0.5705	0.00	5.69	0.018	14.691	518.70	11.802	487.99
9	10.472	-0.722	589.84	595.71	0.00	595.71	0.5514	0.00	8.05	0.029	14.691	518.70	11.968	489.91
10	10.618	-0.879	557.41	569.50	0.00	569.50	0.5259	0.00	11.82	0.043	14.652	518.70	12.155	492.39
11	10.764	-0.898												

** values of parameters on streamlines at station, 8, which is the inlet of rotor number, 1 **

streamline no.	r/rtip	rel. flow angle (deg)	rel. vel. (ft/sec)	rel. mach number	wheel speed (ft/sec)	flow coef.
tip 1.0000	67.15	1387.87	1506.09	1.3927	1387.87	0.4155
1	0.9879	64.37	1328.60	1.3697	1328.60	0.4508
2	0.9457	63.08	1267.79	1.3225	1267.79	0.4567
3	0.9025	61.78	1203.88	1.2711	1203.88	0.4593
4	0.8570	60.39	1136.40	1.2160	1136.40	0.4595
5	0.8089	58.86	1064.27	1.1564	1064.27	0.4577
6	0.7576	57.13	986.14	1.0913	986.14	0.4534
7	0.7020	55.08	900.00	1.0193	900.00	0.4464
8	0.6407	52.53	802.59	0.9378	802.59	0.4357
9	0.5713	49.11	688.01	0.8424	688.01	0.4199
10	0.4898	43.67	543.68	0.7270	543.68	0.3968
11	0.3870					
hub 0.3749						

streamline no.	inc. angle (deg)	ref. angle (deg)	s.s. inc. angle (deg)	in. blade angle (deg)	inlet streamline	element slope (deg)	mach no. at shock location	streamline element sh. loc. as fract of s.s.	cov. chan. margin	min. chk. area	pt. loc. in cov. chan.
1	1.93	1.18	-2.24	-0.93	65.97	-0.82	1.4851	0.6871	0.3129	-0.0050	0.2123
2	8.68	1.23	-2.08	-0.95	63.14	-0.38	1.4480	0.6422	0.3578	0.0011	0.3962
3	15.60	1.86	-1.97	-0.94	61.22	-4.61	1.4005	0.6047	0.3953	0.0083	0.4489
4	22.88	2.41	-1.90	-0.93	59.37	-3.05	1.3575	0.5629	0.4371	0.0084	0.4407
5	30.57	3.12	-1.88	-0.92	57.27	-1.48	1.3172	0.5178	0.4822	0.0073	0.4337
6	38.78	3.88	-1.86	-0.93	54.98	0.24	1.2812	0.4684	0.5316	0.0098	0.4059
7	47.68	4.77	-1.90	-0.94	52.36	2.12	1.2434	0.4157	0.5843	0.0141	0.3670
8	57.49	5.73	-1.94	-0.96	49.35	4.15	1.2109	0.3580	0.6420	0.0259	0.2725
9	68.58	6.77	-2.01	-0.92	45.76	6.46	1.1438	0.2947	0.7053	0.0413	0.2730
10	81.63	7.12	-2.22	-0.87	42.00	9.26	1.0347	0.2296	0.7704	0.0402	0.2532
11	98.06	6.14	-3.03	-1.67	37.53	13.26	0.8731	0.1548	0.7730	0.0544	0.2864

** values of parameters on streamlines at station, 9, which is the outlet of rotor number, 1 **

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

streamline no.	radius in.	axial coord. in.	axial vel. ft/sec	merid. vel. ft/sec	tang. vel. ft/sec	abs. vel. ft/sec	abs. mach no.	abs. flow angle (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.f.)	static press. (psia)	static temp. (deg.f.)
tip	9.756	1.778	521.01	525.67	465.33	702.04	0.5992	41.52	-7.63	0.059	24.816	619.48	580.28
1	9.661	1.793	535.70	537.90	437.69	693.48	0.5961	39.14	-5.19	0.042	24.564	609.93	571.60
2	9.288	1.852	519.70	520.78	429.46	675.01	0.5919	39.51	-3.69	0.030	24.156	604.55	568.20
3	8.903	1.887	518.33	518.74	435.52	677.33	0.5854	40.02	-2.27	0.022	24.039	601.77	565.15
4	8.492	1.942	522.44	522.49	448.80	688.78	0.5969	40.66	-0.84	0.016	24.015	599.98	562.09
5	8.061	2.012	531.48	531.51	476.53	713.85	0.6200	41.88	0.65	0.010	24.141	600.18	559.48
6	7.612	2.105	533.79	534.18	504.52	734.78	0.6398	43.36	2.31	0.002	24.118	599.60	556.46
7	7.138	2.208	537.61	538.85	542.25	764.46	0.6678	45.18	5.89	-0.005	24.127	598.37	552.74
8	6.629	2.331	540.99	543.79	583.50	797.61	0.6993	47.02	8.82	-0.013	24.021	597.42	540.13
9	6.077	2.476	545.14	550.80	641.90	845.82	0.7470	49.37	11.38	-0.024	23.906	597.42	528.03
10	5.457	2.608	552.04	563.11	736.87	927.40	0.8280	52.61					
11	4.728	2.703											
hub	4.664	2.712											

streamline no.	rel. flow angle (deg)	rel. vel. ft/sec	rel. mach number	wheel speed (ft/sec)	flow coef.	head coef.	ideal head coef.	adiab. eff.	diffusion factor	loss coef.	shock loss coef.	element solidity
tip	1.0000	59.35	887.16	1031.20	0.8801	1352.49	0.3709	0.2688	0.3189	0.8430	0.4322	0.1263
1	0.9902	58.05	862.66	1016.63	0.8739	1300.35	0.3813	0.2502	0.2884	0.8674	0.4179	0.1012
2	0.9521	57.48	816.93	968.80	0.8352	1246.39	0.3699	0.2405	0.2712	0.8869	0.4237	0.0859
3	0.9125	55.45	753.33	914.66	0.7905	1188.86	0.3690	0.2380	0.2624	0.9073	0.4363	0.0721
4	0.8704	52.45	679.77	857.37	0.7430	1128.57	0.3719	0.2375	0.2566	0.9255	0.4522	0.0603
5	0.8263	47.94	589.11	735.45	0.6891	1065.64	0.3783	0.2402	0.2573	0.9336	0.4758	0.0576
6	0.7802	42.81	494.75	728.10	0.6340	999.27	0.3800	0.2397	0.2555	0.9384	0.4996	0.0519
7	0.7316	35.61	385.85	662.75	0.5789	928.10	0.3827	0.2399	0.2550	0.9408	0.5239	0.0419
8	0.6795	26.17	267.21	605.90	0.5317	850.71	0.3851	0.2376	0.2515	0.9448	0.5373	0.0338
9	0.6229	12.50	122.11	564.17	0.4983	764.01	0.3880	0.2352	0.2485	0.9463	0.5290	0.0279
10	0.5594	-7.59	-74.39	568.08	0.5072	661.88	0.3930	0.2348	0.2471	0.9502	0.4468	0.0001
11	0.4846											0.0000
hub	0.4780											0.0000

streamline no.	pt. pass.	press. ratio	temp. ratio	local blade forces radius for axial tang.	dev. angle (deg)	out. blade angle (deg)	stream. -- out. blade angle (deg)
1	1.87	1.7359	1.1943	9.7870	20.5025	-12.1113	5.34
2	9.18	1.6750	1.1759	9.3891	18.5443	-11.2246	4.94
3	16.75	1.6442	1.1655	8.9792	16.8583	-10.5517	4.65
4	24.82	1.6363	1.1601	8.5455	15.6210	-10.1409	4.70
5	33.28	1.6346	1.1567	8.0892	14.4054	-9.9099	4.91
6	42.11	1.6432	1.1571	7.6068	13.1701	-9.8967	5.49
7	51.42	1.6416	1.1560	7.0908	11.6515	-9.7118	6.17
8	61.40	1.6422	1.1557	6.5289	9.9680	-9.4948	7.24
9	72.26	1.6350	1.1536	5.9047	8.0075	-9.0424	8.59
10	84.42	1.6272	1.1518	5.1858	5.6512	-8.3995	10.45
11	98.74	1.6261	1.1509	4.3056	2.6904	-8.6693	11.84
1							-19.42

** values of parameters on streamlines at station, 10, which is an annulus **

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip 9.630	3.000												
1	9.575	3.003	568.10	568.64	469.51	737.42	0.6321	39.55	-2.50	0.034	24.803	618.42	19.020
2	9.227	3.025	574.29	574.64	440.60	724.11	0.6244	37.48	-2.01	0.028	24.556	609.68	18.944
3	8.863	3.047	558.01	558.17	431.40	705.45	0.6099	37.70	-1.39	0.023	24.156	604.56	18.851
4	8.475	3.071	555.13	555.16	436.39	706.15	0.6119	38.17	-0.58	0.018	24.041	601.85	18.730
5	8.068	3.096	556.81	556.82	448.43	714.94	0.6210	38.85	0.32	0.013	24.017	600.08	18.576
6	7.641	3.122	562.76	562.90	474.73	736.36	0.6409	40.14	1.29	0.008	24.143	600.29	18.374
7	7.187	3.150	562.36	562.83	501.05	753.54	0.6573	41.68	2.35	0.003	24.120	599.71	18.107
8	6.698	3.180	563.18	564.28	536.72	778.77	0.6812	43.57	3.58	-0.003	24.129	599.57	17.747
9	6.162	3.213	563.06	565.29	575.46	806.66	0.7085	45.51	5.09	-0.010	24.023	598.48	17.249
10	5.560	3.250	564.28	568.59	630.04	848.67	0.7497	47.93	7.06	-0.021	23.908	597.53	16.529
11	4.860	3.293	567.36	577.12	716.77	920.23	0.8208	51.16	10.55	-0.041	23.829	597.10	15.371
hub 4.750	3.300												529.21

1 ** values of parameters on streamlines at station, 11, which is the inlet of stator number, 1, of stage number, 1 **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip 9.590													
1	9.533	4.066	585.76	585.85	471.44	751.98	0.6459	38.82	-0.97	0.013	24.793	617.60	18.797
2	9.196	4.067	595.13	595.17	442.10	741.40	0.6405	36.61	-0.64	0.016	24.549	609.47	18.692
3	8.841	4.069	581.04	581.04	432.47	724.32	0.6273	36.66	-0.12	0.016	24.156	604.57	18.591
4	8.465	4.072	578.40	578.41	436.92	724.89	0.6293	37.07	0.45	0.014	24.042	601.91	18.473
5	8.070	4.074	579.74	579.84	448.32	732.94	0.6378	37.71	1.07	0.010	24.019	600.16	18.327
6	7.635	4.078	585.13	585.40	473.82	753.13	0.6566	38.99	1.75	0.007	24.144	600.37	18.137
7	7.216	4.082	584.21	584.79	499.08	768.80	0.6717	40.48	2.55	0.003	24.121	599.79	17.889
8	6.741	4.087	584.14	585.23	533.25	791.74	0.6935	42.34	3.50	0.000	24.130	599.65	17.559
9	6.222	4.092	582.81	584.73	569.84	816.47	0.7179	44.26	4.65	-0.006	24.024	598.57	17.105
10	5.641	4.099	583.14	586.36	620.94	854.04	0.7549	46.64	6.00	-0.016	23.910	597.62	16.450
11	4.971	4.101	590.85	595.61	700.78	919.70	0.8201	49.64	7.25	-0.039	23.831	597.19	15.382
hub 4.840	4.102												529.38

streamline no. r/rtip	flow coef.	rel. flow angle (deg)
tip 1.0000		
1	0.9933	0.4170
2	0.9579	0.4236
3	0.9210	0.4136
4	0.8818	0.4117
5	0.8407	0.4127
6	0.7975	0.4165
7	0.7517	0.4159
8	0.7023	0.4158
9	0.6482	0.4149
10	0.5877	0.4151
11	0.5179	0.4151
hub 0.5042		-0.46

streamline no. pass.	inc. angle (deg)	ref. angle (deg)	s.s. inc. angle (deg)	in. blade angle (deg)	streamline element element slope (deg)	mach no. at shock (deg)	sh. loc. location of s.s.	cov. chan. as fract of s.s.	min. chk. margin	min. chk. pt. loc. in cov. chan.
1	1.35	12.20	-2.24	5.97	26.63	-0.43	0.9149	0.2453	0.5151	1.0000
2	8.49	13.04	-2.08	6.87	23.57	0.14	0.8825	0.2201	0.5703	1.0000
3	15.94	13.18	-1.97	7.07	23.48	0.58	0.8586	0.2152	0.5952	1.0000
4	23.84	13.16	-1.90	7.08	23.91	1.00	0.8588	0.2123	0.6109	1.0000

TABLE 3 - EXAMPLE OUTPUT DATA SET(with stator resets)

streamline no.	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
5	32.14	13.08	-1.88	7.06	24.63	1.44	0.8692	0.2104	0.6240	0.2350	0.8811			
6	40.85	13.10	-1.88	7.15	25.89	1.98	0.8981	0.2108	0.6332	0.2302	0.8082			
7	50.09	13.13	-1.90	7.25	27.35	2.64	0.9228	0.2108	0.6422	0.2353	0.5963			
8	60.06	13.31	-1.94	7.50	29.03	3.42	0.9594	0.2096	0.6504	0.2445	0.5927			
9	70.96	13.27	-2.01	7.55	30.99	4.29	0.9972	0.2071	0.6608	0.2574	0.3528			
10	83.16	12.98	-2.22	7.36	33.66	5.11	1.0534	0.2033	0.6680	0.2621	0.0000			
11	97.25	12.56	-3.03	7.07	37.07	5.31	1.1521	0.1948	0.6708	0.2572	0.0000			
1	** values of parameters on streamlines at station, 12, which is the outlet of stator number, 1, of stage number, 1 **													
streamline no.	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	9.597	6.336	626.76	626.82	-145.44	643.47	0.5491	-13.06	0.82	0.016	24.024	613.36	19.628	580.41
1	9.518	6.337	641.41	611.51	-120.30	623.23	0.5332	-11.13	1.02	0.009	23.779	608.22	19.647	577.27
2	9.281	6.339	588.26	588.35	-103.92	597.46	0.5117	-10.02	1.01	0.000	23.436	604.52	19.651	576.06
3	8.884	6.340	580.85	580.97	-98.17	589.20	0.5053	-9.59	1.17	-0.003	23.326	602.22	19.643	574.53
4	8.504	6.342	579.14	579.34	-95.04	587.08	0.5040	-9.32	1.49	-0.003	23.289	600.62	19.627	573.12
5	8.147	6.344	585.97	586.31	-94.44	593.87	0.5101	-9.15	1.94	-0.003	23.365	600.82	19.612	572.69
6	7.754	6.345	580.52	581.08	-92.79	588.45	0.5054	-9.07	2.52	-0.004	23.271	600.25	19.594	572.63
7	7.340	6.347	573.98	574.85	-94.05	582.50	0.5001	-9.29	3.17	-0.005	23.167	600.11	19.575	573.05
8	6.876	6.349	557.56	558.82	-94.39	566.73	0.4864	-9.59	3.84	-0.006	22.930	599.06	19.548	573.44
9	6.391	6.351	529.73	531.28	-100.87	540.77	0.4636	-10.75	4.38	-0.008	22.559	598.15	19.510	574.83
10	5.843	6.353	471.74	473.20	-116.32	487.28	0.4164	-13.81	4.49	-0.004	21.884	597.73	19.457	578.80
11	5.181	6.353												
hub	5.043	6.354												
streamline no.	r/rtip	flow coef.	head coef.	ideal head coef.	stator po.ratio	stage po.ratio	stage ad.eff.	diffusion factor	stator loss coef.	shock loss coef.	element solidity			
tip	1.0000	0.4462	0.2519	0.3189	0.9690	1.6805	0.7898	0.4625	0.1283	0.0000	1.2896			
1	0.9918	0.4352	0.2333	0.2884	0.9686	1.6215	0.8091	0.4433	0.1315	0.0000	1.3356			
2	0.9588	0.4187	0.2250	0.2712	0.9702	1.5952	0.8294	0.4417	0.1294	0.0000	1.3878			
3	0.9237	0.4135	0.2225	0.2624	0.9702	1.5877	0.8482	0.4427	0.1286	0.0000	1.4481			
4	0.8862	0.4123	0.2217	0.2566	0.9696	1.5852	0.8639	0.4427	0.1282	0.0000	1.5174			
5	0.8469	0.4132	0.2234	0.2573	0.9677	1.5904	0.8681	0.4468	0.1297	0.0000	1.5974			
6	0.8059	0.4171	0.2234	0.2553	0.9648	1.5840	0.8664	0.4610	0.1364	0.0000	1.6918			
7	0.7628	0.4132	0.2213	0.2555	0.9648	1.5840	0.8664	0.4610	0.1364	0.0000	1.6918			
8	0.7165	0.4086	0.2190	0.2550	0.9601	1.5769	0.8589	0.4820	0.1455	0.0000	1.8070			
9	0.6660	0.3969	0.2138	0.2515	0.9545	1.5608	0.8498	0.5122	0.1581	0.0000	1.9525			
10	0.6089	0.3771	0.2055	0.2485	0.9435	1.5355	0.8269	0.5612	0.1811	0.0005	2.1463			
11	0.5398	0.3358	0.1915	0.2471	0.9183	1.4936	0.7748	0.6504	0.2304	0.0073	2.4278			
hub	0.5255													
streamline no.	pos.	local blade forces	radius for axial tang. (in.)	lbs/in	dev. angle (deg)	out. stream. dev. angle (deg)								
1	1.72	9.5267	2.3036	12.5891	13.02	-26.08								
2	8.66	9.1984	1.9452	11.1844	11.33	-22.46								
3	16.09	8.8524	1.8651	10.0780	10.60	-20.62								
4	23.99	8.4844	1.8665	9.5151	10.17	-19.77								
5	32.28	8.0983	1.9164	9.2222	9.85	-19.17								
6	40.91	7.6944	2.0805	9.1769	9.71	-18.86								
7	50.00	7.2678	2.1879	8.9677	9.59	-18.67								
8	59.74	6.8088	2.3256	8.7651	9.57	-18.86								
9	70.39	6.3068	2.4268	8.3748	9.49	-19.08								
10	82.44	5.7422	2.5138	7.9126	9.62	-20.37								
11	96.98	5.0759	2.4395	8.2169	10.43	-24.24								

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

** values of parameters on streamlines at station, 13, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. flow vel. (ft/sec)	abs. mach no.	abs. angle stream. (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.f.)	static press. (psia)	static temp. (deg.f.)
tip 9.603	7.340	600.19	600.31	-145.20	617.62	0.5261	-13.60	-1.16	24.012	613.03	19.940	582.68
1 9.534	7.340	601.75	601.83	-120.16	613.71	0.5247	-11.29	-0.93	23.771	608.09	19.757	578.09
2 8.212	7.340	591.85	591.88	-103.81	600.91	0.5148	-9.95	-0.58	23.434	604.50	19.608	575.71
3 8.874	7.340	592.91	592.91	-98.01	600.96	0.5139	-9.39	0.01	23.325	602.24	19.503	573.44
4 8.518	7.340	595.88	595.92	-94.79	603.42	0.5197	-9.04	0.73	23.288	600.66	19.433	571.61
5 9.148	7.340	604.62	604.83	-94.06	612.10	0.5285	-8.84	1.51	23.364	600.86	19.393	570.97
6 7.765	7.340	598.84	599.34	-92.25	606.40	0.5216	-8.75	2.34	23.271	600.29	19.381	570.95
7 7.362	7.340	589.67	590.59	-93.31	597.92	0.5140	-8.98	3.20	23.167	600.16	19.395	571.63
8 6.931	7.340	567.79	569.24	-93.41	576.85	0.4955	-9.32	4.09	22.933	599.11	19.436	572.56
9 6.438	7.340	530.39	532.37	-99.55	541.60	0.4643	-10.59	4.95	22.566	598.21	19.507	574.81
10 5.920	7.340	456.61	458.95	-114.57	473.03	0.4038	-14.02	5.78	21.904	597.79	19.609	579.95
11 5.259	7.340											
hub 5.131	7.340											

** values of parameters on streamlines at station, 14, which is the inlet of rotor number, 2 **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. flow vel. (ft/sec)	abs. mach no.	abs. angle stream. (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.f.)	static press. (psia)	static temp. (deg.f.)
tip 9.499	8.707	685.89	690.63	-146.80	706.06	0.6063	-12.00	-6.71	23.997	612.58	18.782	572.86
1 9.430	8.703	685.26	687.79	-121.18	698.38	0.6016	-9.99	-4.91	23.761	607.92	18.664	569.03
2 9.134	8.686	671.43	672.46	-104.39	680.52	0.5870	-8.82	-3.18	23.431	604.47	18.613	567.52
3 8.824	8.665	666.29	666.51	-98.25	673.72	0.5819	-8.39	-1.48	23.324	602.27	18.599	566.04
4 8.498	8.640	662.23	662.23	-94.71	668.97	0.5783	-8.14	0.21	23.287	600.71	18.620	564.99
5 8.156	8.612	662.55	662.89	-93.66	669.47	0.5786	-8.04	1.85	23.363	600.91	18.675	565.13
6 7.798	8.581	648.50	649.67	-91.54	656.08	0.5667	-8.02	3.43	23.271	600.35	18.768	565.99
7 7.420	8.547	629.78	632.22	-92.23	638.91	0.5511	-8.30	5.03	23.168	600.21	18.889	567.63
8 7.012	8.509	597.41	601.47	-91.96	608.46	0.5239	-8.69	6.66	22.935	599.18	19.071	569.83
9 6.560	8.465	548.90	554.77	-97.56	563.29	0.4837	-9.97	8.34	22.574	598.28	19.278	572.96
10 6.041	8.415	467.96	474.76	-111.65	487.71	0.4167	-13.23	9.71	21.925	597.86	19.489	578.89
11 5.397	8.337											
hub 5.265	8.322											

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

inlet streamline										streamline element										streamline element									
streamline no.	inc. angle (deg)	ref. angle (deg)	s.s. inc. angle (deg)	inc. angle (deg)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	static press. (psia)	static temp. (deg. r.)														
tip	9.374	9.729	555.27	557.43	334.22	649.94	0.5151	30.95	-5.04	0.084	38.575	711.32	32.315	678.43															
1	9.311	9.733	528.78	529.77	364.12	642.84	0.5116	34.50	-3.51	0.054	38.479	704.70	32.308	672.48															
2	9.048	9.749	511.48	511.89	401.07	650.30	0.5188	38.08	-2.27	0.032	38.588	702.10	32.242	669.11															
3	8.767	9.772	505.62	505.69	423.04	659.31	0.5273	39.91	-0.98	0.015	38.700	699.54	32.148	665.61															
4	8.471	9.798	505.35	505.36	439.97	670.05	0.5373	41.04	0.36	-0.002	38.687	696.84	31.913	661.77															
5	8.164	9.827	504.53	504.76	458.52	681.93	0.5476	42.25	1.72	-0.018	38.716	696.23	31.705	659.89															
6	7.845	9.859	499.11	499.83	486.50	697.51	0.5610	44.23	3.08	-0.032	38.614	695.79	31.318	657.76															
7	7.509	9.897	490.56	492.06	519.00	715.19	0.5758	46.53	4.48	-0.046	38.526	696.20	30.907	656.22															
8	7.150	9.941	471.70	474.25	559.12	733.16	0.5914	49.69	5.95	-0.060	38.206	695.71	30.293	653.68															
9	6.758	9.993	441.80	441.80	594.55	740.73	0.5986	53.38	7.45	-0.074	37.411	693.96	29.497	651.03															
10	6.311	10.056	351.69	356.34	608.52	705.18	0.5706	59.65	9.27	-0.073	35.298	688.28	28.424	649.34															
11	5.747	10.143																											
hub	5.647	10.158																											
** values of parameters on streamlines at station, 15, which is the outlet of rotor number, 2 **																													
streamline no.	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	static press. (psia)	static temp. (deg. r.)																
tip	9.374	9.729	555.27	557.43	334.22	649.94	0.5151	30.95	-5.04	0.084	38.575	711.32	32.315	678.43															
1	9.311	9.733	528.78	529.77	364.12	642.84	0.5116	34.50	-3.51	0.054	38.479	704.70	32.308	672.48															
2	9.048	9.749	511.48	511.89	401.07	650.30	0.5188	38.08	-2.27	0.032	38.588	702.10	32.242	669.11															
3	8.767	9.772	505.62	505.69	423.04	659.31	0.5273	39.91	-0.98	0.015	38.700	699.54	32.148	665.61															
4	8.471	9.798	505.35	505.36	439.97	670.05	0.5373	41.04	0.36	-0.002	38.687	696.84	31.913	661.77															
5	8.164	9.827	504.53	504.76	458.52	681.93	0.5476	42.25	1.72	-0.018	38.716	696.23	31.705	659.89															
6	7.845	9.859	499.11	499.83	486.50	697.51	0.5610	44.23	3.08	-0.032	38.614	695.79	31.318	657.76															
7	7.509	9.897	490.56	492.06	519.00	715.19	0.5758	46.53	4.48	-0.046	38.526	696.20	30.907	656.22															
8	7.150	9.941	471.70	474.25	559.12	733.16	0.5914	49.69	5.95	-0.060	38.206	695.71	30.293	653.68															
9	6.758	9.993	441.80	441.80	594.55	740.73	0.5986	53.38	7.45	-0.074	37.411	693.96	29.497	651.03															
10	6.311	10.056	351.69	356.34	608.52	705.18	0.5706	59.65	9.27	-0.073	35.298	688.28	28.424	649.34															
11	5.747	10.143																											
hub	5.647	10.158																											
streamline no.	r/rtip	rel. flow angle (deg)	rel. tang. vel. (ft/sec)	rel. vel. (ft/sec)	rel. mach number	wheel speed (ft/sec)	flow coeff.	head coeff.	ideal head (ft)	adiab. eff. factor	diffusion loss	shock loss	shock element solidity																
tip	1.0000	60.10	969.33	1118.18	0.8862	1303.54	0.4175	0.3012	0.3559	0.8463	0.4235	0.1113	0.0468	1.3075															
1	0.9933	59.59	902.56	1046.55	0.8329	1266.68	0.3976	0.3039	0.3484	0.8722	0.4441	0.0954	0.0405	1.3486															
2	0.9652	58.22	826.26	971.98	0.7754	1227.34	0.3846	0.3134	0.3613	0.8924	0.4724	0.0852	0.0363	1.3934															
3	0.9352	56.46	762.90	915.28	0.7320	1185.94	0.3802	0.3173	0.3498	0.9073	0.4932	0.0763	0.0335	1.4441															
4	0.9037	54.29	702.98	865.78	0.6943	1142.94	0.3800	0.3174	0.3435	0.9166	0.5098	0.0693	0.0310	1.5014															
5	0.8709	51.73	639.73	814.89	0.6544	1098.26	0.3794	0.3156	0.3426	0.9219	0.5300	0.0688	0.0287	1.5668															
6	0.8368	48.49	564.73	754.16	0.6065	1051.24	0.3753	0.3164	0.3429	0.9225	0.5570	0.0721	0.0269	1.6428															
7	0.8010	44.41	481.98	688.73	0.5546	1000.98	0.3669	0.3177	0.3449	0.9211	0.5896	0.0785	0.0260	1.7342															
8	0.7627	39.21	386.53	612.09	0.4937	946.07	0.3547	0.3184	0.3468	0.9179	0.6308	0.0892	0.0264	1.8484															
9	0.7209	33.19	286.96	527.91	0.4266	883.50	0.3394	0.3145	0.3437	0.9150	0.6782	0.1019	0.0216	2.0005															
10	0.6732	28.83	196.11	406.74	0.3291	804.63	0.2645	0.2951	0.3246	0.9091	0.7559	0.1216	0.0131	2.2239															
11	0.6131																												
hub	0.6024																												

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

streamline no. radius	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	9.315	11.010	582.62	336.17	672.78	0.5344	29.98	-1.31	0.010	38.565	710.52	31.883	675.26
1	9.257	11.010	560.55	365.85	669.48	0.5339	33.13	-1.24	0.004	38.471	704.52	31.813	669.56
2	9.005	11.010	546.92	402.38	679.00	0.5429	36.34	-0.81	0.006	38.579	702.03	31.702	666.05
3	8.738	11.010	544.80	423.62	690.12	0.5532	37.87	-0.32	0.003	38.690	699.54	31.559	662.35
4	8.459	11.010	541.28	439.64	697.33	0.5604	39.08	0.22	-0.001	38.677	696.92	31.384	658.91
5	8.170	11.010	542.21	457.14	709.24	0.5707	40.13	0.83	-0.005	38.706	696.32	31.169	656.99
6	7.869	11.010	530.99	483.81	718.49	0.5788	42.33	1.52	-0.011	38.604	695.87	30.901	655.51
7	7.551	11.010	520.93	514.75	732.65	0.5907	44.64	2.28	-0.015	38.517	696.29	30.555	654.31
8	7.209	11.010	495.15	552.88	742.69	0.5995	48.11	3.16	-0.021	38.201	695.80	30.100	652.66
9	6.834	11.010	454.59	586.03	742.42	0.6000	52.13	4.18	-0.025	37.427	694.07	29.478	650.94
10	6.403	11.010	366.62	597.17	701.70	0.5675	56.32	5.74	-0.047	35.412	688.54	28.581	649.99
11	5.857	11.010											
hub	5.763	11.010											

** values of parameters on streamlines at station, 16, which is an annulus **

streamline no. radius	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	9.297	11.887	604.32	337.07	692.00	0.5507	29.15	-0.62	0.025	38.556	710.02	31.511	672.71
1	9.232	11.888	580.59	366.61	686.69	0.5484	32.27	-0.75	0.020	38.462	704.39	31.483	667.60
2	8.986	11.892	565.81	402.92	694.64	0.5561	35.45	-0.55	0.008	38.569	701.97	31.397	664.31
3	8.726	11.897	563.50	423.85	705.11	0.5659	36.95	-0.24	0.003	38.680	699.53	31.261	660.70
4	8.455	11.900	560.14	439.47	711.96	0.5728	38.12	0.15	-0.001	38.667	696.96	31.092	657.34
5	8.173	11.903	561.13	456.46	723.36	0.5827	39.13	0.58	-0.005	38.696	696.37	30.883	655.46
6	7.880	11.907	550.39	482.44	731.90	0.5902	41.24	1.10	-0.008	38.594	695.93	30.628	654.04
7	7.572	11.913	540.15	512.41	744.71	0.6010	43.48	1.75	-0.009	38.508	696.34	30.307	652.97
8	7.242	11.919	514.45	545.12	752.45	0.6079	46.87	2.61	-0.006	38.195	695.86	29.901	651.57
9	6.881	11.926	472.37	580.14	748.77	0.6035	50.79	3.74	-0.002	37.442	694.15	29.365	650.27
10	6.467	11.937	392.12	587.35	706.63	0.5716	56.22	3.55	-0.035	35.529	688.78	28.588	649.69
11	5.954	11.970											
hub	5.843	11.977											

** values of parameters on streamlines at station, 17, which is the inlet of stator number, 2 **

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

streamline no.	pct. pass.	local blade forces		--- out. stream. ---	
		radius (in.)	tang. (lbs/in)	dev. angle (deg)	out. blade angle (deg)
1	1.62	9.2400	1.3775	7.1196	11.14
2	9.30	8.9871	1.8445	7.2331	10.36
3	17.09	8.7249	2.3538	7.4581	9.92
4	25.00	8.4551	2.6091	7.5111	9.61
5	33.12	8.1767	2.7692	7.4998	9.43
6	41.52	7.8880	2.9129	7.4736	9.37
7	50.30	7.5851	3.1354	7.5066	9.43
8	59.60	7.2626	3.3780	7.4775	9.57
9	69.60	6.9426	3.6721	7.2928	9.83
10	80.93	6.5140	3.7622	6.3946	10.62
11	97.40	5.9785	2.6597	5.1590	30.05

** values of parameters on streamlines at station, 19, which is an annulus **

streamline no.	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream slope (deg)	stream curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	9.303	14.440	541.94	541.94	5.16	541.97	0.4277	0.55	-0.10	-0.004	37.842	707.57	33.466	684.71
1	9.248	14.440	541.12	541.12	23.32	541.63	0.4286	2.47	0.14	-0.001	37.861	703.62	33.465	680.77
2	8.991	14.440	548.04	548.04	37.23	549.32	0.4355	3.89	0.39	0.002	38.014	701.58	33.467	678.07
3	8.729	14.440	553.08	553.08	44.31	554.88	0.4407	4.58	0.60	0.005	38.139	699.46	33.475	675.45
4	8.463	14.440	558.06	558.06	47.49	559.55	0.4439	4.93	0.80	0.007	38.138	697.20	33.489	673.34
5	8.190	14.440	560.78	560.78	49.09	563.05	0.4401	5.09	0.99	0.008	38.161	696.65	33.505	672.78
6	7.909	14.440	561.57	561.57	50.05	564.99	0.4328	5.28	1.17	0.008	38.021	696.22	33.523	673.13
7	7.614	14.440	562.49	562.49	49.53	564.93	0.4252	5.31	1.31	0.007	37.877	696.63	33.540	674.30
8	7.302	14.440	569.18	569.18	47.87	571.57	0.4062	5.37	1.38	0.006	37.499	696.17	33.555	675.75
9	6.967	14.440	569.33	569.33	41.24	551.57	0.3598	5.21	1.34	0.004	36.633	694.58	33.566	678.51
10	6.585	14.440	532.49	532.49	232.61	77.71	245.25	18.47	1.85	-0.012	34.439	689.99	33.573	685.30
11	6.033	14.440	232.49	232.49										
hub	5.947	14.440												

** values of parameters on streamlines at station, 20, which is an annulus **

streamline no.	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream slope (deg)	stream curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	9.298	15.700	568.37	568.37	5.16	568.41	0.4494	0.52	-0.40	0.006	37.826	707.21	33.033	682.06
1	9.243	15.705	566.45	566.45	23.31	566.92	0.4493	2.36	0.35	0.016	37.846	703.47	33.049	678.43
2	8.995	15.728	570.70	570.70	37.17	572.00	0.4541	3.73	1.01	0.023	37.996	701.50	33.086	675.99
3	8.742	15.751	572.42	572.42	44.20	574.34	0.4567	4.41	1.58	0.028	38.119	699.44	33.142	673.71
4	8.485	15.774	566.75	566.75	47.31	569.09	0.4531	4.77	2.06	0.032	38.118	697.24	33.212	671.97
5	8.221	15.798	562.50	562.50	48.86	565.14	0.4500	4.96	2.47	0.035	38.141	696.70	33.293	671.77
6	7.946	15.823	548.90	548.90	49.76	551.81	0.4392	5.17	2.81	0.038	38.002	696.27	33.383	672.50
7	7.658	15.849	534.79	534.79	49.20	537.81	0.4275	5.25	3.06	0.040	37.860	696.67	33.481	674.11
8	7.351	15.877												

TABLE 3 - EXAMPLE OUTPUT DATA SET (with statior resets)

9	7.019	15.907	506.04	506.84	47.51	509.06	0.4042	5.36	3.22	0.042	37.491	696.22	31.586	676.01
10	6.638	13.842	443.38	444.70	40.91	446.58	0.3538	5.26	3.26	0.043	36.666	694.67	33.693	679.11
11	6.084	13.392	229.57	229.86	77.06	242.43	0.1912	18.53	2.88	0.045	34.648	690.23	33.796	685.65
hub	6.000	16.000												

** values of parameters on streamlines at station, 21, which is an annulus **

streamline no.	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	9.319	17.000	635.46	637.00	5.15	637.02	0.5060	0.46	3.99	0.109	37.811	706.84	31.866	675.22
1	9.266	17.010	619.42	621.68	23.19	622.11	0.4949	2.14	4.88	0.101	37.829	703.32	32.114	673.14
2	9.040	17.053	611.03	614.02	36.90	615.12	0.4898	2.14	5.63	0.094	37.979	701.41	32.347	671.90
3	8.808	17.097	601.96	605.63	43.77	607.21	0.4839	4.13	6.31	0.089	38.101	699.41	32.573	670.65
4	8.568	17.143	587.22	591.50	46.75	593.35	0.4731	4.52	6.90	0.085	38.100	697.28	32.793	669.80
5	8.319	17.191	574.74	579.60	48.17	581.60	0.4636	4.75	7.42	0.083	38.123	696.75	33.006	670.34
6	8.059	17.240	553.81	559.12	48.96	561.26	0.4469	5.00	7.91	0.081	37.985	696.32	33.217	671.73
7	7.784	17.293	532.69	538.41	48.29	540.57	0.4298	5.13	8.35	0.081	37.844	696.72	33.425	673.92
8	7.490	17.349	503.32	510.67	46.53	505.47	0.4012	5.28	8.77	0.082	37.486	696.28	33.634	676.35
9	7.168	17.410	430.08	435.67	39.96	437.50	0.3464	5.24	9.19	0.086	36.697	694.75	33.838	679.82
10	6.796	17.481	218.79	222.21	75.01	234.53	0.1849	18.65	10.06	0.102	34.833	690.47	34.032	686.18
11	6.250	17.585												
hub	6.171	17.600												

** values of parameters on streamlines at station, 22, which is an annulus **

streamline no.	radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	9.406	17.750	647.18	656.55	5.10	656.57	0.5224	0.45	9.69	0.135	37.802	706.63	31.511	673.03
1	9.355	17.764	626.99	636.94	22.93	637.36	0.5076	2.06	10.14	0.122	37.821	703.23	31.840	671.55
2	9.141	17.825	614.95	625.66	36.43	626.72	0.4984	3.33	10.62	0.113	37.969	701.36	32.137	670.72
3	8.920	17.888	602.71	614.18	43.15	615.70	0.4909	4.02	11.09	0.107	38.091	699.40	32.417	669.82
4	8.691	17.953	585.22	597.36	46.01	599.13	0.4779	4.40	11.57	0.102	38.090	697.30	32.686	669.28
5	8.453	18.020	570.20	583.06	47.32	584.98	0.4654	4.64	12.06	0.098	38.112	696.77	32.940	670.06
6	8.204	18.091	546.98	560.37	47.99	562.43	0.4479	4.90	12.55	0.096	37.975	696.35	33.190	671.66
7	7.941	18.165	523.80	537.74	47.23	539.81	0.4252	5.02	13.08	0.094	37.836	696.75	33.429	674.02
8	7.658	18.245	430.08	435.15	45.38	435.17	0.3993	5.17	13.64	0.094	37.482	696.31	33.664	676.56
9	7.349	18.333	224.14	232.84	72.35	243.82	0.1922	17.26	15.70	0.088	34.940	690.62	34.072	685.98
10	6.993	18.434												
11	6.480	18.579												
hub	6.406	18.600												

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

** values of parameters on streamlines at station, 23, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	9.572	18.500											
1	9.523	18.519	627.35	649.60	5.01	649.62	0.5166	0.44	15.04	37.794	706.42	31.626	673.53
2	9.318	18.599	609.67	632.55	22.50	632.95	0.5040	2.04	15.45	37.812	703.14	31.909	671.90
3	9.106	18.681	598.53	622.21	35.69	623.23	0.4965	3.28	15.85	37.959	701.31	32.189	671.01
4	8.889	18.766	586.52	610.97	42.19	612.43	0.4882	3.95	16.27	38.080	699.38	32.465	670.12
5	8.663	18.854	569.02	594.06	44.90	595.75	0.4751	4.32	16.69	38.079	697.32	32.734	669.62
6	8.426	18.946	553.95	579.73	46.07	581.58	0.4635	4.54	17.16	38.102	696.80	32.988	670.40
7	8.177	19.043	530.66	556.95	46.61	558.50	0.4450	4.78	17.67	37.965	696.37	33.238	672.00
8	7.910	19.147	507.48	534.42	45.73	536.38	0.4283	4.89	18.27	37.827	696.78	33.475	674.33
9	7.619	19.260	471.02	498.09	43.77	500.01	0.3967	5.02	18.97	37.479	696.34	33.708	676.84
10	7.285	19.390	403.82	431.54	37.27	433.14	0.3429	4.94	19.88	36.736	694.85	33.929	680.22
11	6.813	19.574	223.52	240.57	68.81	250.22	0.1973	15.96	21.70	35.047	690.77	34.130	685.89
hub	6.745	19.600											

** values of parameters on streamlines at station, 24, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	9.812	19.250											
1	9.766	19.274	596.46	639.77	4.89	639.79	0.5085	0.44	21.20	37.785	706.21	31.791	674.31
2	9.573	19.374	576.61	618.77	21.90	619.16	0.4925	2.03	21.27	37.803	703.04	32.140	673.15
3	9.374	19.478	565.70	607.51	34.67	608.50	0.4843	3.27	21.38	37.949	701.25	32.434	672.38
4	9.169	19.585	555.25	597.03	40.90	598.43	0.4766	3.92	21.56	38.069	699.37	32.695	671.43
5	8.957	19.695	539.92	581.64	43.42	583.26	0.4648	4.27	21.83	38.068	697.35	32.935	670.80
6	8.735	19.810	527.44	569.71	44.44	571.44	0.4552	4.46	22.21	38.090	696.83	33.146	671.34
7	8.503	19.931	506.72	549.34	44.82	551.16	0.4386	4.66	22.72	37.955	696.40	33.353	672.70
8	8.257	20.060	486.27	529.79	43.80	531.60	0.4224	4.73	23.37	37.828	696.37	33.541	674.76
9	7.991	20.198	452.87	496.79	41.73	498.54	0.3955	4.80	24.27	37.476	696.37	33.727	676.99
10	7.689	20.355	392.72	435.26	35.31	436.69	0.3457	4.84	25.34	36.757	694.90	33.904	680.03
11	7.280	20.568	232.94	264.76	64.40	272.48	0.2150	13.67	28.38	35.158	690.94	34.070	685.15
hub	7.219	20.600											

** values of parameters on streamlines at station, 25, which is an annulus **

streamline no. radius (in.)	axial coord. (in.)	axial vel. (ft/sec)	merid. vel. (ft/sec)	tang. vel. (ft/sec)	abs. vel. (ft/sec)	abs. mach no.	abs. flow angle (deg)	stream. slope (deg)	stream. curv. (1./in.)	total press. (psia)	total temp. (deg.r.)	static press. (psia)	static temp. (deg.r.)
tip	10.149	20.000											
1	10.106	20.027	518.76	580.55	4.72	580.57	0.4597	0.47	26.68	37.776	706.01	32.785	679.76
2	9.916	20.146	512.37	570.00	21.14	570.39	0.4523	2.12	25.99	37.792	702.95	32.946	677.60
3	9.721	20.268	512.48	568.09	33.43	569.07	0.4518	3.37	25.56	37.939	701.20	33.082	675.95
4	9.523	20.393	511.23	565.84	39.38	567.21	0.4509	3.98	25.38	38.058	699.35	33.204	674.26
5	9.328	20.508	503.64	557.55	41.71	559.11	0.4448	4.28	25.41	38.057	697.37	33.321	672.98

TABLE 3 - EXAMPLE OUTPUT DATA SET (with stator resets)

6	9.111	20.651	498.27	552.66	42.61	554.30	0.4410	4.41	25.63	0.040	38.079	696.85	33.415	672.88
7	8.894	20.788	483.95	538.76	42.85	540.46	0.4298	4.55	26.07	0.038	37.945	696.43	33.513	673.64
8	8.666	20.930	469.82	526.08	41.74	527.73	0.4192	4.54	26.74	0.037	37.809	696.83	33.592	675.11
9	8.424	21.092	443.13	500.55	39.59	502.11	0.3984	4.52	27.71	0.037	37.473	696.41	33.673	676.74
10	8.155	21.258	392.24	449.28	33.29	450.52	0.3569	4.24	29.19	0.038	36.778	694.96	33.746	679.13
11	7.823	21.465	260.22	308.57	60.00	314.35	0.2483	11.00	32.51	0.041	35.268	691.10	33.821	683.40
hnd	7.756	21.500												

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13. ABSTRACT (Maximum 200 words) An off-design axial-flow compressor code is presented and is available from COSMIC for predicting the aerodynamic performance maps of fans and compressors. Steady axisymmetric flow is assumed and the aerodynamic solution reduces to solving the two-dimensional flow field in the meridional plane. A streamline curvature method is used for calculating this flow-field outside the blade rows. This code allows for bleed flows and the first five stators can be reset for each rotational speed; capabilities which are necessary for large multi-stage compressors. The accuracy of the off-design performance predictions depend upon the validity of the flow loss and deviation correlation models. These empirical correlations for the flow loss and deviation are used to model the real flow effects and the off-design code will compute through small reverse flow regions. The input to this off-design code is fully described and a user's example case for a two-stage fan is included with complete input and output data sets. Also, a comparison of the off-design code predictions with experimental data is included which generally shows good agreement.				
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