

5/10/00
322
1-4

NASA WIRING FOR SPACE APPLICATIONS PROGRAM

Norman Schulze
Office of Space Access and Technology
Spacecraft Systems Division
NASA Headquarters
Washington, DC



NASA Code XS Overview

- ORGANIZATION
- CHARTER
- PROGRAMS

NASA Strategic Plan

- Enterprises
 - Aeronautics
 - Mission to Planet Earth
 - Space Science
 - Human Exploration
 - Space Technology
- Function
 - Communications
 - Human Resources
 - Physical Resources



Mission

**Pioneer, With Industry, the Development and Use of
Space Technology to Secure National Economic
Competitiveness, Promote Industrial Growth and to
Support Space Missions**

Space Technology Enterprise Goals



1. Reduce the Cost of Access to Space

- Reusable Launch Vehicle
- Expendable Launch Vehicle
- In-Space Transportation

2. Provide Innovative Technologies to Enable Ambitious, Future Space Missions (ITP)

- Spacecraft Systems (Power, Propulsion, Structures, etc.)
- Instrument Technologies
- Operations

3. Build Capability in the U. S. Space Industry Through Focused Space Technology Efforts

- Communications
- Remote Sensing
- Space Processing

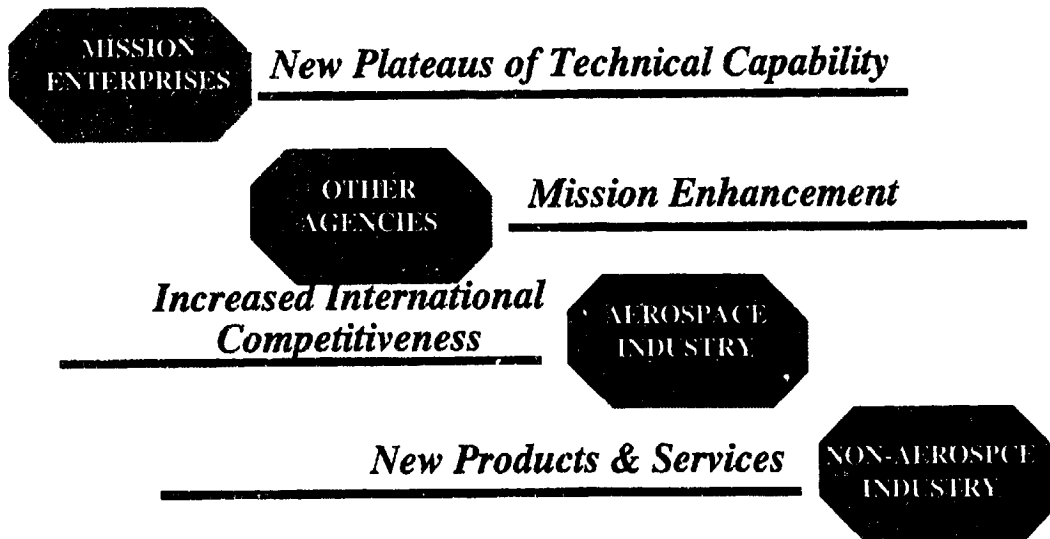
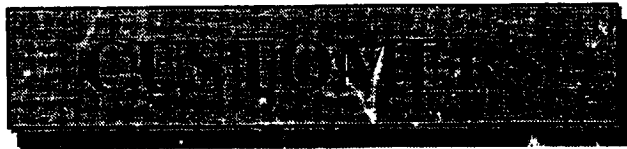
4. Share the Harvest of Space Technology with the U.S. Industrial Community

- Technology Transfer - "Agenda for Change" (New Way of Doing Business)

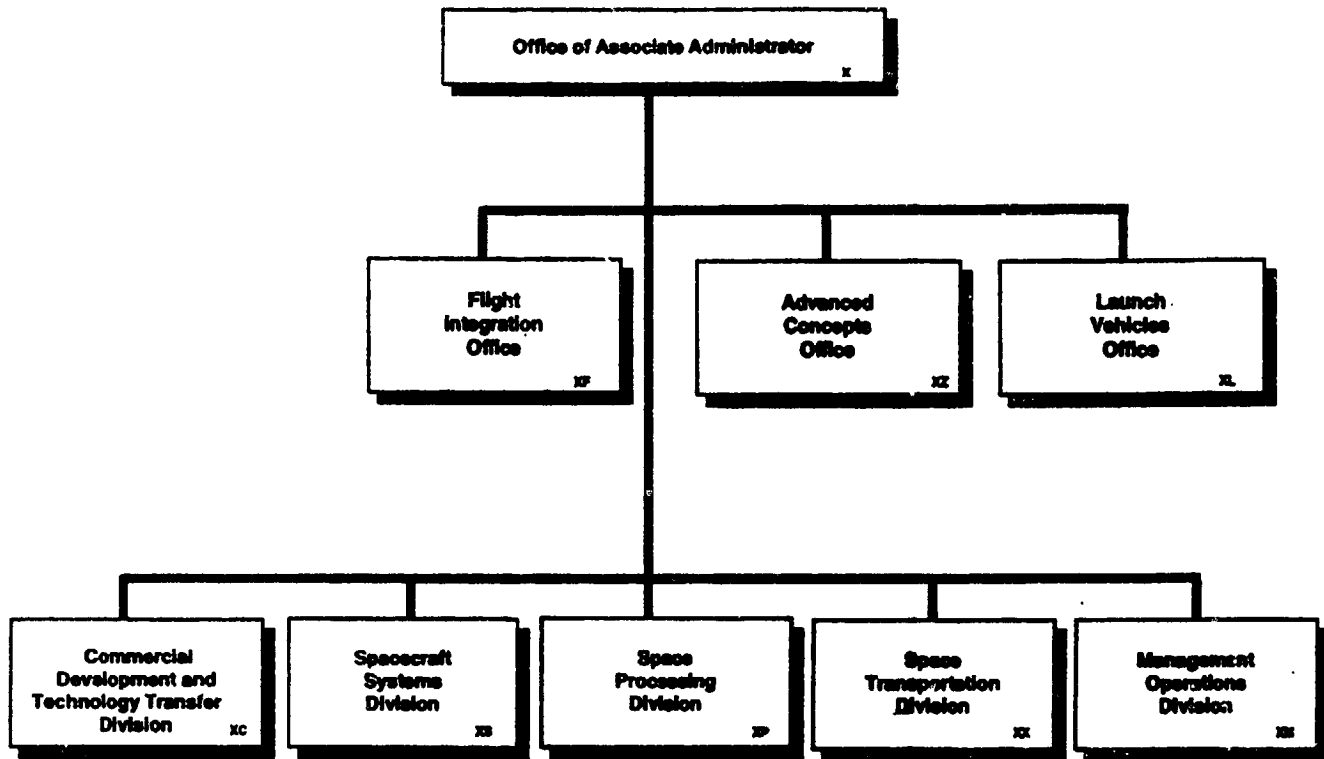


Operating Principles !

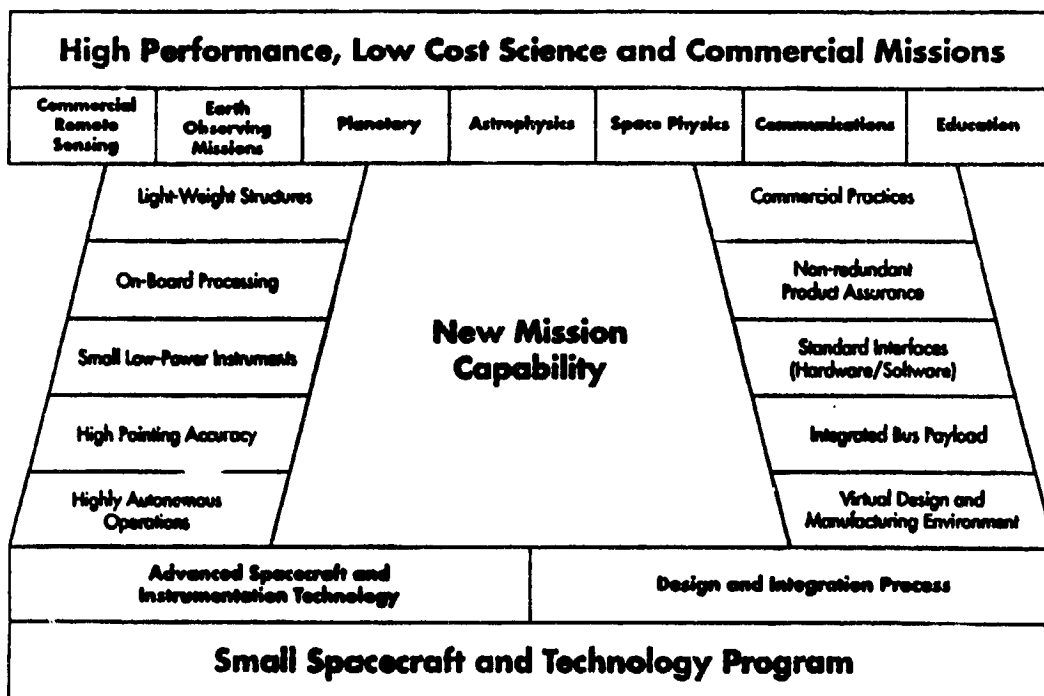
- Meet the Customers Needs
- Work With Industry
- Reduce the Cost of Access to Space
- Commercialization of Space Is Essential to NASA
- Commercialization and Technology Transfer Is Everybody's Job
- Consider Commercialization at Technology Program Initiation
- Effectively Use Space Station



OFFICE OF SPACE ACCESS AND TECHNOLOGY



Foundation for Future Missions



NEW MILLENNIUM / ESSP

CHALLENGES

- **Replace large, multi-instrument spacecraft with multiple small single instrument “sciencecraft”**
 - Change focus from “instruments on a spacecraft” to “the instrument is the spacecraft”
- **Return information, not data**
- **Wide, unconstrained interaction with users and information distribution to users**
- **Low initial cost, low operations cost**

Small Spacecraft Tech Initiative

• Lewis

– Payloads

- **Hyper Spectral (30M, 358 Bands)**
- **UV cosmic measurement**
- **Cloud detection/editing**

– 20 Technology Demonstrations

- **Integrated Thermal/Structural Design**
- **Advanced power concepts**
- **Autonomous on-orbit maintenance**
- **Advanced C&DH and data bus arch.**
- **Data compression**

• Clark

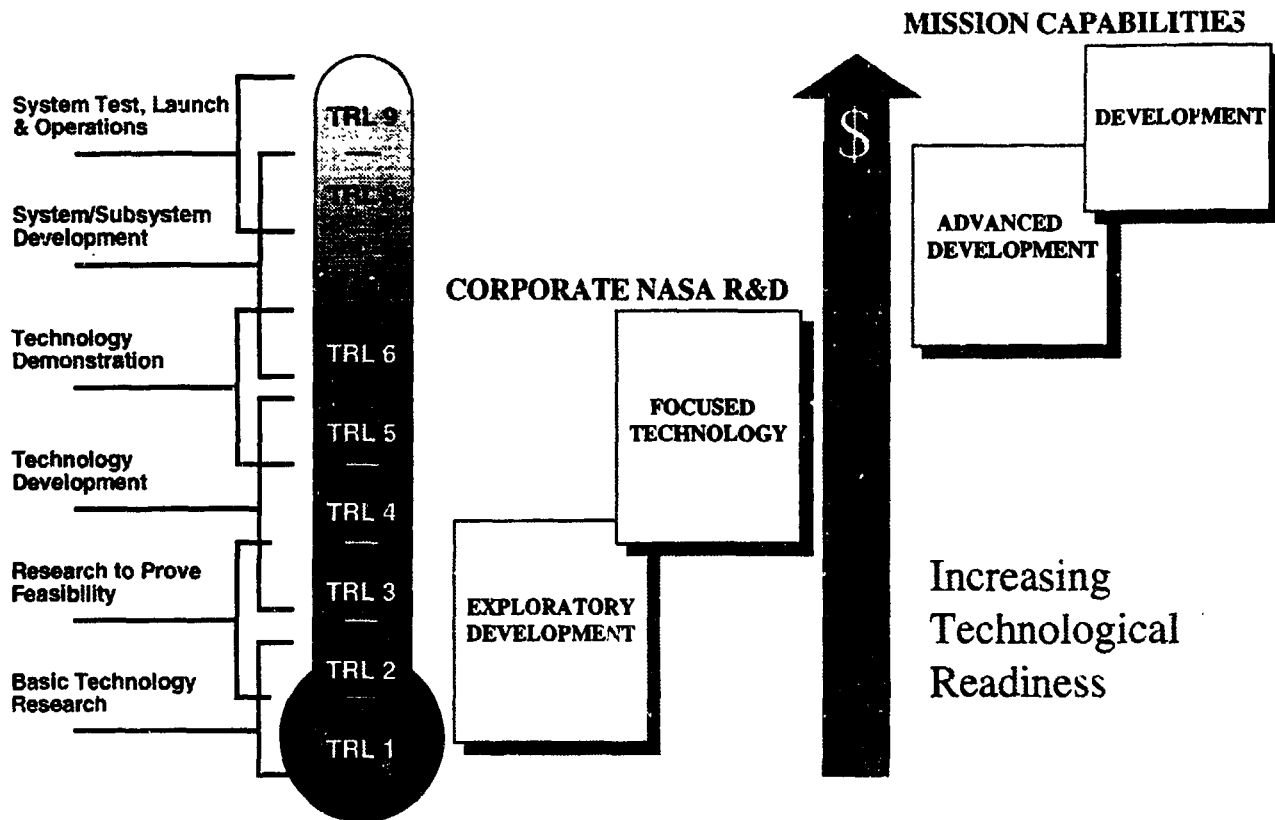
– Payloads

- **3-meter panchromatic (world view)**
- **C₀ detection in atmosphere**
- **Room temp. X-Ray detectors**

– 36 Technology Demonstrations

- **Advanced attitude control**
- **Advanced photovoltaic concepts**
- **Advanced power management and distribution**
- **On-board processing**
- **No shock release devices**

Technology Readiness Levels (TRLs)



TECHNOLOGY ENTERPRISE STATUS

Access to Space

- National Space Transportation Policy Issued, NASA Implementation Plan Approved by OSTP & OMB
- Cooperative Agreements for X-33 & X-34 Signed With Industrial Partners
- 2000 Hour Ground Test of Ion Flight Experiment Prototype Thruster Successfully Completed

Innovative Technologies

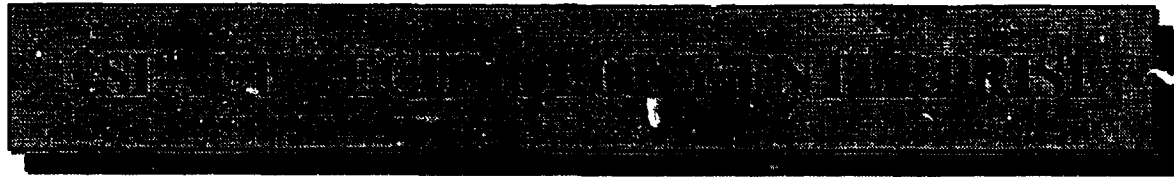
- Parallel Contracts for SSTI Awarded, spacecraft construction started, launch date established and launch vehicle selection completed.
- New Millennium Spacecraft Technology Program Defined With Codes S & Y
- Mars Pathfinder Micro-Rover Fabrication Nearing Completion

Space Applications

- ACTS Fully Operational
- Commercial Remote Sensing Program Has Leveraged \$38.5M of Industry Funding, and Led to the Creation of 25 New Products, Over 140 New Jobs, and Revenues Exceeding \$66M and Is Developing Hyperspectral Capability Which Will Enable Movement Into New Markets
- Starting Large Animal Trials on Diabetes Treatment, Based on in-Space Developed Technology of Microencapsulation

Technology Transfer

- Agenda for Change Plan Approved, Agency-Wide Team Established, Performance Measurement Metrics Collected, and Technology Transfer Principle Added to NASA Strategic Plan



SUMMARY IMPLEMENTATION STRATEGY!

- **Develop Technology in Cooperation With and Responsive to User Requirements, With Upfront Consideration of Dual Use**
- **Proactively Transfer Technology to NASA Missions, Other Agencies and Aerospace and Non-Aerospace Industries**

Background

SPACE MISSIONS WITH ELECTRICAL WIRING SYSTEM FAILURE

Mission	Cause	Result
Gemini 8	Electrical Wiring Short	Shortened Mission - Near Loss of Crew
Apollo 204	Damaged Insulation, Electrical Spark, 100% O ₂	Fire, 3 Astronauts Lost
Apollo 13	Damaged Insulation/Short Circuit/Flawed Design	Oxygen Tank Explosion, Mission Incomplete
STS - 6	Abrasion of Insulation/Arc Tracking	Wire Insulation Pyrolysis 6 Conductors Melted
STS - 28	Damaged Insulation/Arc Tracking	Teleprinter Cable Insulation Pyrolysis
Magellan	Wrong Connection, Wiring Short	Wiring Insulation Pyrolysis - Ground Processing
Spacelab	Damaged Insulation/Arc Tracking	Wiring Insulation Pyrolysis During Maintenance
Delta 178/GOES-G	Mechanical or Electrochemical Insulation Damage	Loss of Vehicle
ESA - Olympus	Electrical Wiring Short	Loss of Solar Array

NASA Wiring for Space Applications Program

- **OBJECTIVES:**

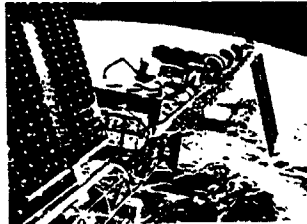
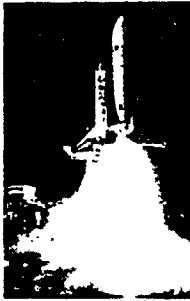
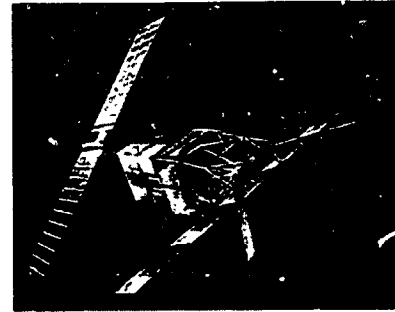
- Improve safety, performance, and reliability of wiring systems for space applications
- Develop improved wiring technologies for NASA flight programs and commercial applications

ELECTRICAL POWER WIRING PROGRAM

GOAL: TO PROVIDE A TECHNOLOGY BASE FOR THE DEVELOPMENT OF LIGHTWEIGHT, ARC TRACK-RESISTANT AND RELIABLE WIRING SYSTEMS FOR AEROSPACE APPLICATIONS.

APPROACH

- IDENTIFY MISSION REQUIREMENTS AND APPLICATION ENVIRONMENTS
- EVALUATE POTENTIAL WIRING SYSTEMS AND ESTABLISH A DATABASE
- INVESTIGATE ADVANCED TECHNOLOGIES RELEVANT TO WIRING FAILURE PREVENTION, DETECTION, AND ISOLATION.
- ESTABLISH GUIDELINES AND RECOMMENDATIONS

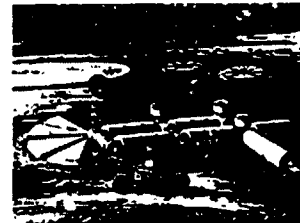


TECHNOLOGICAL DEVELOPMENTS

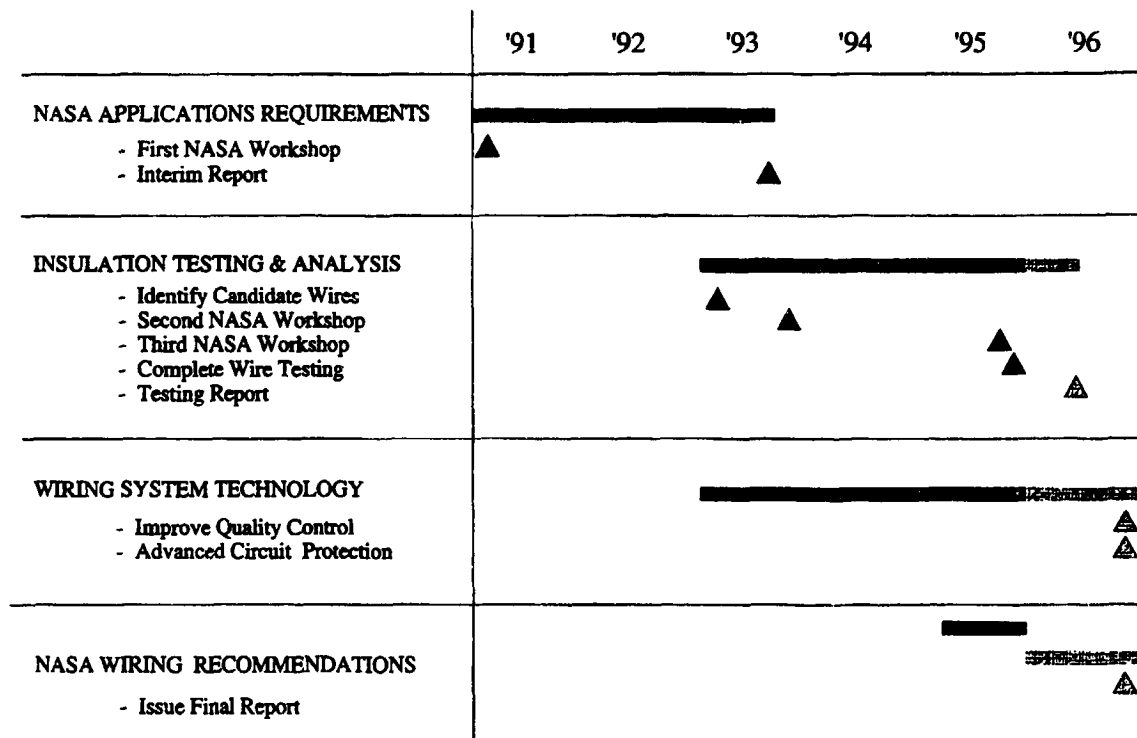
- NEW INSULATING MATERIALS
- NEW WIRING CONSTRUCTIONS
- IMPROVED SYSTEM DESIGN
- ADVANCED CIRCUIT PROTECTION

APPLICATIONS

- PRESSURIZED MODULES
- TRANS-ATMOSPHERIC VEHICLES
- LEO/GEO ENVIRONMENTS
- LUNAR AND MARTIAN ENVIRONMENTS



NASA Wiring for Space Applications Program



- **ACCOMPLISHMENTS:**

- First Workshop, July 1991:

- Wiring system operational experience
- NASA wiring requirements
- Wire manufacturing technologies

Proceedings: "First NASA Workshop on Wiring for Space Applications", NASA CP-10145, July, 1994

- Interim Report, June 1993:

- NASA spacecraft environments
- NASA unique testing requirements
- Related wiring programs

Report: "Operational Environments for Electrical Power Wiring on NASA Spacecraft", NASA TM-106655, June 1994

- Second Workshop, July 1993:

- Program overviews: NASA, AF, NAWC, ESA
- Space wiring failures
- Candidate wiring constructions
- New wiring insulation
- Test methodology and standardization

Proceedings: "Second NASA Workshop on Wiring for Space Applications", NASA CP-3244, October, 1993

NASA Wiring for Space Applications Program

- **R & D PROGRAMS:**

- System design
- Candidate wiring constructions
- New insulating materials
- Protection techniques
- Quality control

- **ORGANIZATIONS:**

- NASA
- DOD laboratories
- FAA
- Aerospace Industry
- ESA
- Academia
- Technical committees

Tests Performed vs. Wiring Constructions Matrix.

Test	Construction	MIL-W-81381/7	MIL-W-81381/11	MIL- 22759/12	MIL- 2759/16	MIL-W-22759/18	MIL-W-22759/32	MIL-W-22759/34	MIL-W-22759/43	MIL-W-16878	SSQ-21652	SSQ-21656	MP571-0086	Fibrex - TKT	Tensolite - TKT	Thermatics - TKT	Gore HS-725	Nyma #3 - TKT	Barcel - TKT	Nema #2 - TKT	DuPont (P-FP)	Brand Rex - TKT	Champlain - TKT	TRW - PFPI
Examine Product		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Workmanship		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Wire Wall Thickness		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Conductor Diameter		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Finished Diameter		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Finished Weight		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Wire Surface Marability		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Impulse Dielectc		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Insulation Resistance		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Spark Test		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Dry Dielectc Test		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Voltage Withstand		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Dielectc Constani		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
CI/CEV (AC & DC)		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Surface Resistance		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Time/Current to Smoke		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Wire Fusing Time		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Dry Arc Resistance		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
BSI Dry Arc Resistance		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Arc Tracking - SSF		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Arc Tracking - NHB Method		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Arc Tracking - MIL-W-2223		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Dielectc Strength		○	●	○	●	●	●	●	●	●	●	●	●	○	○	○	○	○	○	○	○	○	○	○
Abrasion		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Dynamic Cut Through		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Flex Life		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Notch Propagation		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Stiffness and Spnngback		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Crush Resistance		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Insulation Impact Resistance		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Tensile Strength		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Wire to Wire Rub		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Aging Stability		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Thermal Index		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Thermal Shock		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Thermal Aging		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Cold Bend		○	○	○	●	●	●	●	●	●	●	●	●	○	○	○	○	○	○	○	○	○	○	○
Thermal Cycling		○	○	○	●	●	●	●	●	●	●	●	●	○	○	○	○	○	○	○	○	○	○	○
Flammability - Aircraft		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Toxicity - Burning		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Smoke Quantity		●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●

Tests Performed vs. Wiring Constructions Matrix (Cont'd).

Test / Construction	MIL-W-81381/7	MIL-W-81381/11	MIL-W-22759/12	MIL-W-22759/16	MIL-W-22759/18	MIL-W-22759/32	MIL-W-22759/34	MIL-W-22759/43	MIL-W-16878	SSQ-21652	SSQ-21656	MP571-0086	Fibrex - TKT	Tensolite - TKT	Thermatics - TKT	Gore HS-725	Nema #3 - TKT	Barcol - TKT	Nema #2 - TKT	DuPont (P-FPI)	Brand Rex - TKT	Champlan - TKT	TRW - PFPI
Flame Propagation	○	●	●	●	●	●	●	●	●	●	●	●	○	○	○	○	●	●	●	●	●	●	○
Flash Point of Liquids	○	●	●	●	●	●	●	●	●	●	●	●	○	○	○	○	●	●	●	●	●	●	○
Wire Flammability	○	●	●	●	●	●	●	●	●	●	●	●	○	○	○	○	●	●	●	●	●	●	○
Odor Assessment	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Offgassing	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Arc Tracking - 30% O ₂	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Wet Arc Tracking (ASTM)	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Hydrolysis (ASTM)	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Wet Arc Tracking (SAE)	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Fluid Immersion	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Forced Hydrolysis (SAE)	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Humidity Resistance	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Wicking	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Impact of LOX and GOX	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●	●
Fluid Compatibility	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Gas Compatibility	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Weight Loss	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
QIV/CEV - Vacuum	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Outgassing (VCM)	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Arc Tracking - Vacuum	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Weathering Resistance	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
VUV Exposure	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
VUV/AC Exposure	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Radiation Exposure	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
AO Exposure	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Arc Tracking - μ g	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Arc Tracking - μ g	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Flame Spread Rate - μ g	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Corona Discharge - Plasma	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Debris Impacts	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○
Electrostatic Dust	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○	○

Key:

- Tests performed by DOD programs [11 - 13]
- Some DOD testing, more necessary [11 - 13]
- Tests performed by NASA programs [26, 56, 57]
- Some NASA testing, more necessary [26, 56, 57]
- Tests not required for this program
- Additional tests to be performed

NASA Wiring for Space Applications Program

- **ACTIVITIES:**

- Third Workshop, July 1995:
 - Program status: NASA, AF, NAWC, FAA, ESA
 - Wiring test results
 - Advancements in materials and constructions
 - New system topologies
- Final Report, 1996:
 - Comprehensive test results
 - Recommendations and guidelines
- Transfer Technology to NASA Flight Programs and Aerospace Industry

- **CONCLUSIONS:**

- Wiring system failures in space and commercial applications have shown the need for arc track resistant wiring constructions
- Preliminary data indicates the performance of the Tensolite and Filotex hybrid constructions are the best of the various candidates
- One construction will be recommended after comprehensive evaluation and analysis of all testing data
- Detailed presentations of the test efforts and results to date will follow

Wiring Workshop Charge

Determine next steps for:

- s/c wiring
- new wiring advances
- circuit protection
- improvement in quality control measures