

PLANET-B

NAG5-2918

101132
1N-33-012
OCIT
67538

FINAL REPORT

Technical Notes
and
Drawings

July 25, 1996

**George R. Carignan
Space Physics Research Laboratory
University of Michigan
2455 Hayward St.
Ann Arbor MI 48109-2143**

24-Aug-95

To: K. Arnett, B. Block, J. Eder, L. Frost, S. Battel

From: J. Maurer

Subj: Planet-B NMS FB $\pm 75V$ Bias Transformer

The design of the transformer designated as T101 (061-0351) in the Filament/Bias module (061-0119) in the Planet-B NMS instrument was verified because of the differences from the GCMS and INMS instrument designs. A breadboard of a representation of the Hybrid 2301065, Bias Drive A driving a 2N3700 NPN transistor, with dual 75V secondaries, with loads, was used to test the circuit. See attached drawing 061-0356A.

The initial transformer design that was wound with bifilar secondaries was too unstable to test. The second 1408 transformer with a split bobbin and the feedback winding below the primary was also found to be unstable. (It was nearly impossible to keep the circuit from squeeging).

The third transformer tested was wound as shown on attached drawing 061-0351, Rev —, Date 24-Aug-95. This transformer has the feedback on the outside of the resonant winding. The primary goal of the design was to have as tight of magnetic coupling as possible to the resonant winding, and as loose of coupling as possible to the primary. Further, the circuit AC ground is connected to the winding at the feedback end of the secondary winding. See drawing.

This transformer proved to be very stable – it is virtually impossible to make this design squeeg.

An emitter resistor (R129A) was added to this circuit, as referenced to the GCMS design, to protect Q102 from thermal runaway in the event of a turnon with a non-resonant circuit or load short. This was verified to protect Q102 for at least 30 seconds in the event of a short. Approximately 1% of the 41mW input power is lost in this protection resistor under normal operation.

The circuit was verified to operate normally when a radiated Q102 (2N3700), (low Beta) transistor was substituted for the normal 2N3700. It should be noted that the monitored drive voltage went to approximately 2.7V with this low gain transistor.

Waveforms

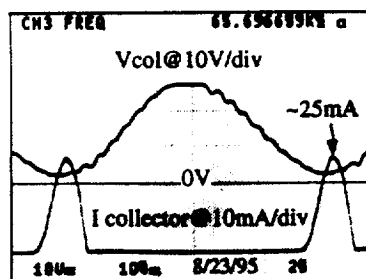


Figure 1

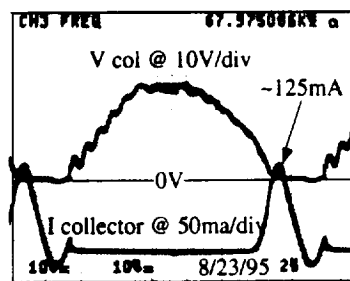


Figure 2

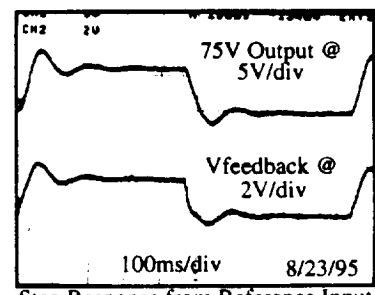


Figure 3

Figure 1 shows the collector voltage and current at a controlled normal output of 75 volts. This is about 90% of maximum voltage output. The waveforms in Figure 2 are the same measurements, but with a maximum voltage output where the control loop overdrives the output transistor. Note the significantly greater current peak.

Figure 3 is the result of a small step applied to the input reference voltage. The ringing is due to the time constants that are designed into the hybrid that cannot be changed for this circuit.

TECHNICAL NOTES

Measured Values

Frequency with 3300pF cap:	72KHz
Input Current at 15.0V	2.73 mA
Output voltage:	75V
Load Resistance:	1.5M (All windings)
Common Mode noise to gnd, 7-8:	.03mv AC RMS (Control/Anode)
Output noise, 7-8:	.8mv AC RMS
Common Mode noise to gnd, 5-6:	.2mv AC RMS (Monitor/Emission)

Attachments

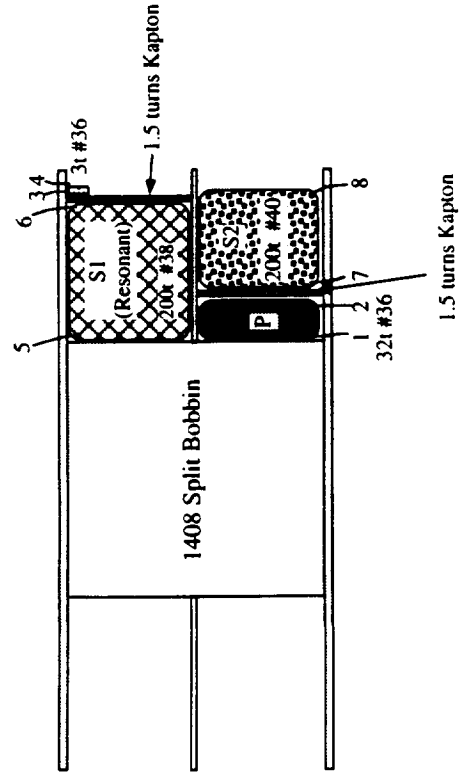
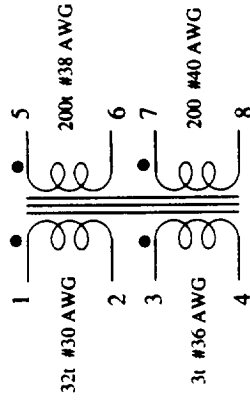
Dwg. 061-0356, Rev. A, 24-Aug-95, Pg. 5.38

Dwg. 061-0351, Rev. -, 24-Aug-95, Pg. 5.22

LIST OF MATERIALS

Item	QTY	P/N	Description
1	1	I408PA250-3B7	Philips Magnetic Core
2	1	I408F1D	Philips Plastic Bobbin
3	AR	EPON 815/V140	Encapsulate

NOTES:



Space Physics Research Laboratory
College of Engineering
University of Michigan

FB +/-75V Bias Transformer
Filament Bias Supply
Planet B - NMS

061-NMS 0TK63 J0351-FB +/-75 Bias Xfmr. 061-0351

Printed: 8/24/94 1:11 PM

Drawing Page

5.22

22-Aug-95

To: K. Arnett, B. Block, J. Eder, L. Frost

From: J. Maurer

Subj: Filament Current Transformer Phasing

Overview

The S/Ns EM1, EM2, and FM1 Filament/Bias (FB) module current transformers were assembled with the phasing such that the positive (monitored) output occurred during the non-modulated (non-controlled) portion of the 50 kHz AC input. The DC value of the current monitor was observed to make a significant change in the DC value for a change in the filament voltage.

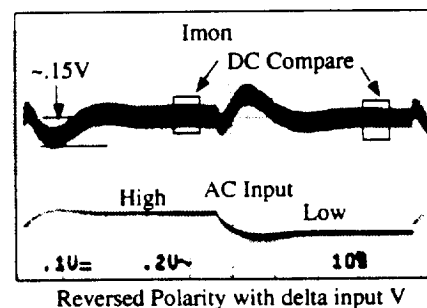
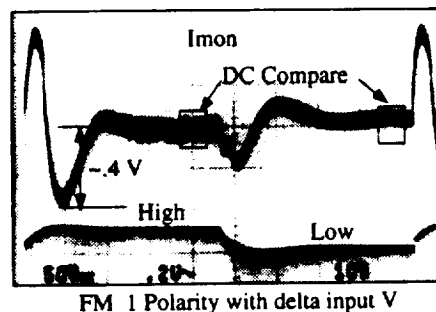
Measurements made on the GCMS KIWI breadboard unit confirmed that the breadboard had the same characteristics. When the phasing of the current transformer was reversed the voltage sensitivity was reduced by a factor of 4 or 5. Further, the negative voltage spike from the output of the current measurement circuit causing a shut down was approximately 1/2 as great when the phasing was reversed.

With either phasing the sensitivity to a change in the AC voltage and the unwanted spike can be significantly reduced by selecting the R101 trim resistor. With the "reversed" phasing the voltage sensitivity can be reduced to near zero.

It was established that a change in the input AC power caused about the same effects as a change in the filament load voltage. A test procedure was established using a 5% modulation of the AC input voltage and monitoring the Imon output through the AMUX channel to select the best value for R101. The nominal value is about 18K which is approximately 20% greater than the sensitivity (positive) trim resistor.

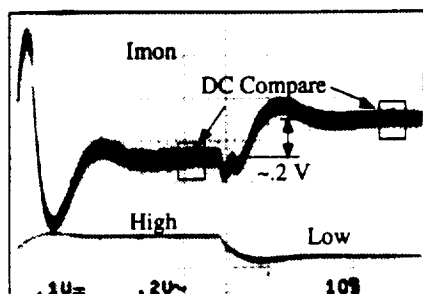
Test Results

An unpotted, flight, filament magnetic assembly was used to verify the results from the KIWI magnetic assemblies. The following pictures were taken to illustrate typical waveforms.

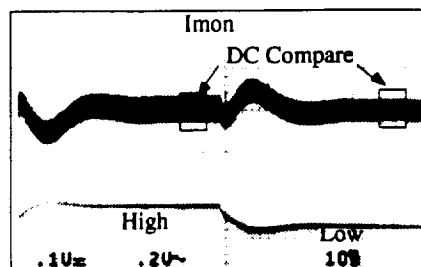


The above pictures show that the DC level shift with the "reversed" polarity is much less. Also that the spikes are significantly less in the "reversed" assembly. While an AC change of this magnitude is not expected in the finished system, the reduced sensitivity to noise spikes could prove to be significant.

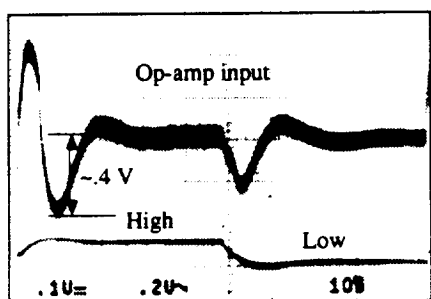
The current monitor is sensitive to the PWM off time. This is illustrated by changing the average AC input voltage. When the average AC input was increased by 1 volt, the sensitivities to a delta AC change is shown below.



FM 1 Polarity with 22 v input (high)

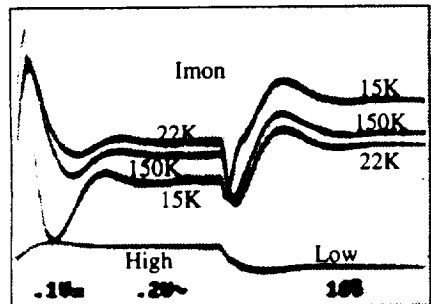


Reversed Polarity with 22 v input (high)



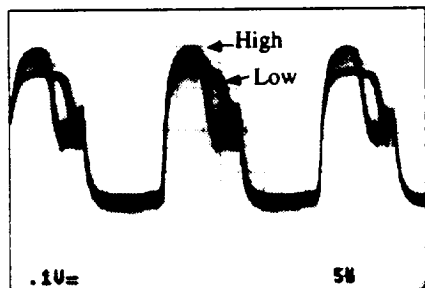
FM 1 Polarity input to op-amp

This picture was taken to show that the voltage through the coupling capacitor to the op-amp used to control the turnon current is the same as that measured on the plus side of the current monitor. The output of the AMUX was confirmed to be the same as these pictures except that the high frequency noise is filtered.



FM1 Polarity with delta R101 trim

This picture was taken as an aid in understanding the sensitivity to the R101 trim resistor. This resistor is a trim for the negative output of the Imon transformer. As can be seen if the R101 value is too low the voltage output is lower when the AC is high. If the trim value is too big of value, the voltage output is higher when the AC is high. It should be noted that this relationship is not linear and different AC deltas and average amplitudes will change the required value.

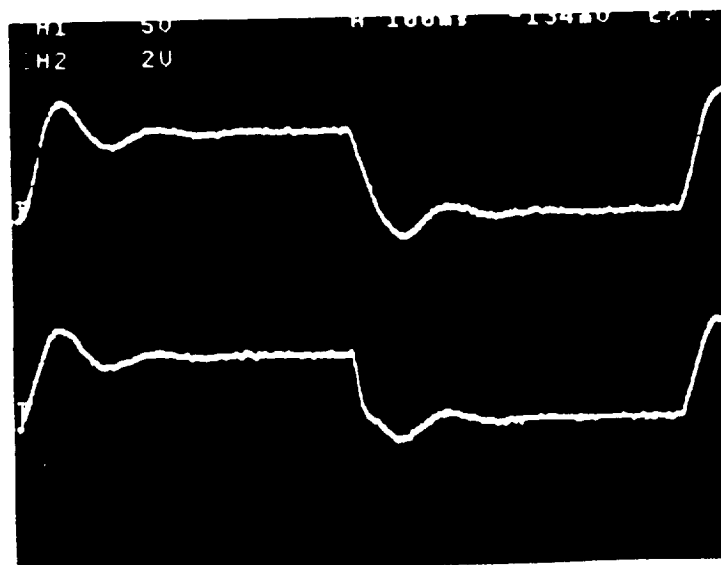
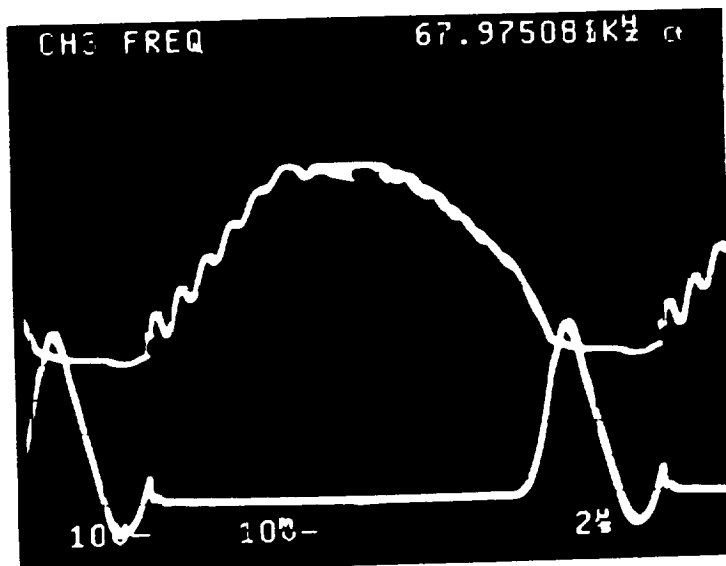
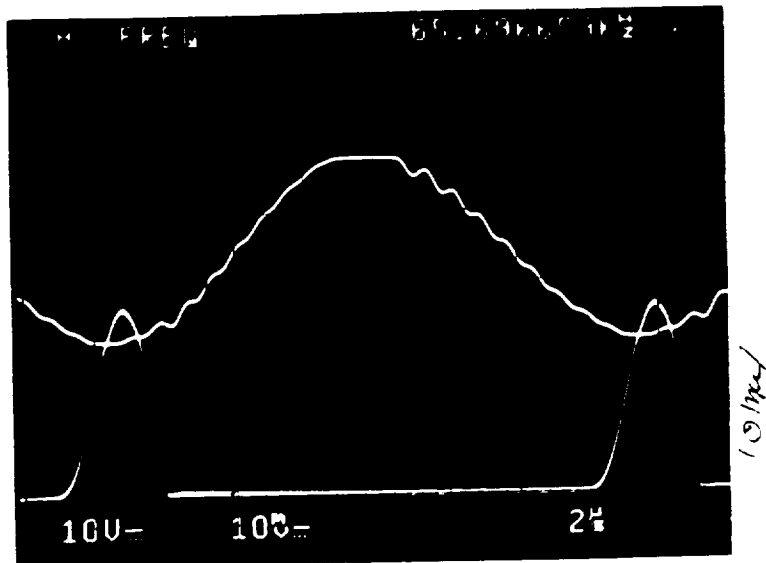


Filament Voltage 19.64 & 22.20 V Input

The AC amplitude was selected to cause an off time of 50% (of the positive or controlled half cycle), changing to about a 10% off time. This is consistent to the values selected and measured during the ion source tests at GSFC.

This picture shows the filament voltage change with the FM1 phasing and a 5% change in AC input.

061-3050
ju



061-3050



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THE UNIVERSITY OF MICHIGAN
SPACE PHYSICS RESEARCH LABORATORY
COLLEGE OF ENGINEERING
DEPARTMENT OF ATMOSPHERIC, OCEANIC AND SPACE SCIENCES

061-3051



313-936-7775
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9102407554 (TELEX)

August 30, 1995

Mr. Steven H. Feng
Code 915.0
NASA Goddard Space Flight Center
Greenbelt, MD 20771

SUBJECT: Interface Requirements for Planet-B NMS RF/DC Supply

Dear Steve:

Please find enclosed a set of schematic drawings for the Control Board of the RF/DC Supply. We are building the breadboard version of the RF/DC to these prints. The interface to your digital logic is straightforward, as you can see.

The RF/DC Supply requires two sixteen-bit serial words to configure (these could be thirteen-bit words if you find that easier to implement). The AC word contains the twelve-bit AC amplitude code and a single bit denoting low or high frequency. The DC word contains the twelve-bit DC amplitude code. The thirteenth bit is ignored. The MSB of both AC and DC codes is shifted first. Each word is latched by a positive-going clock pulse on the appropriate enable line after each data word is shifted in. A low-going reset pulse loads zeroes into both the shift registers and latches. I would in principle send you a timing diagram, but I think that the interface is easy and the drawings speak for themselves.

To summarize:

AC Word

Bits 0 through 11 are AC amplitude data (MSB is bit 11);
Bit 12 is frequency datum (1=low frequency, 0= high frequency);
Bits 13 through 15 are not used (if sixteen bits are shifted);
Bit 15 is shifted first.

DC Word

Bits 0 through 11 are DC amplitude data (MSB is bit 11);
Bits 12 through 15 are not used (if sixteen bits are shifted);
Bit 15 is shifted first.

AC and DC Enables

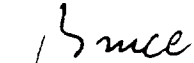
Each data latch requires a positive-going enable after each data word is shifted into the input shift register.

Reset

A low level resets (and holds at reset) both the input shift registers and all data latches.

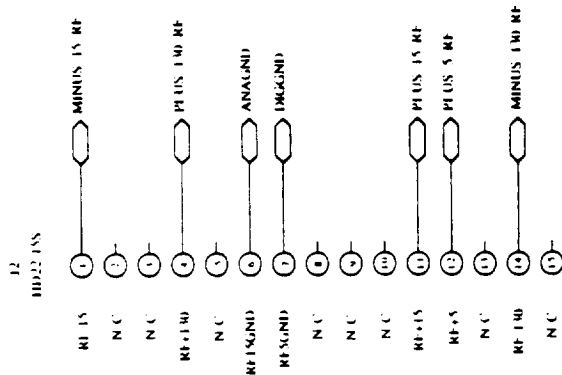
I hope this is sufficient to define the interface for you. Please do not hesitate to call if you have questions.

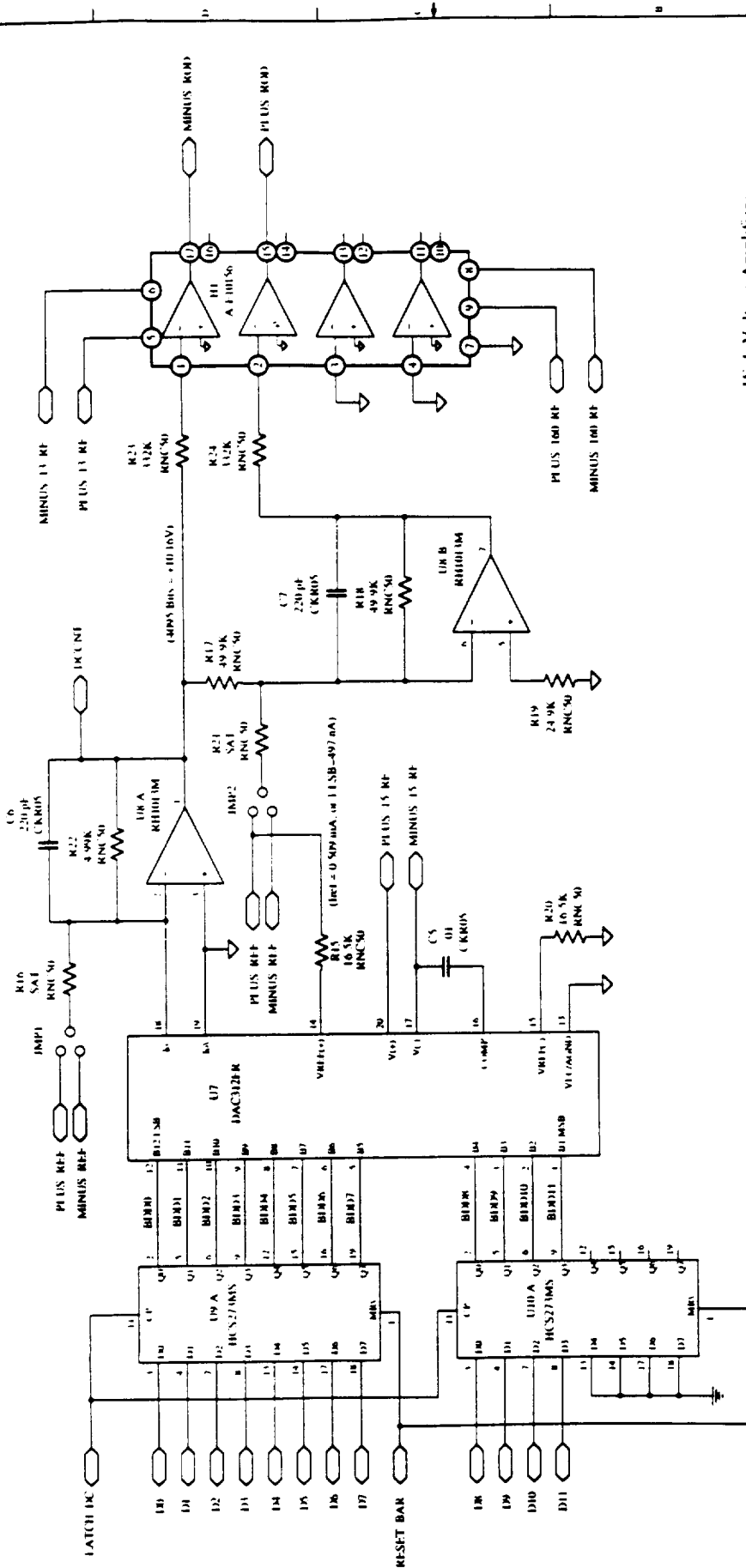
Best regards,



Bruce Block
Senior Engineer in Research

Enclosure

[illegible]

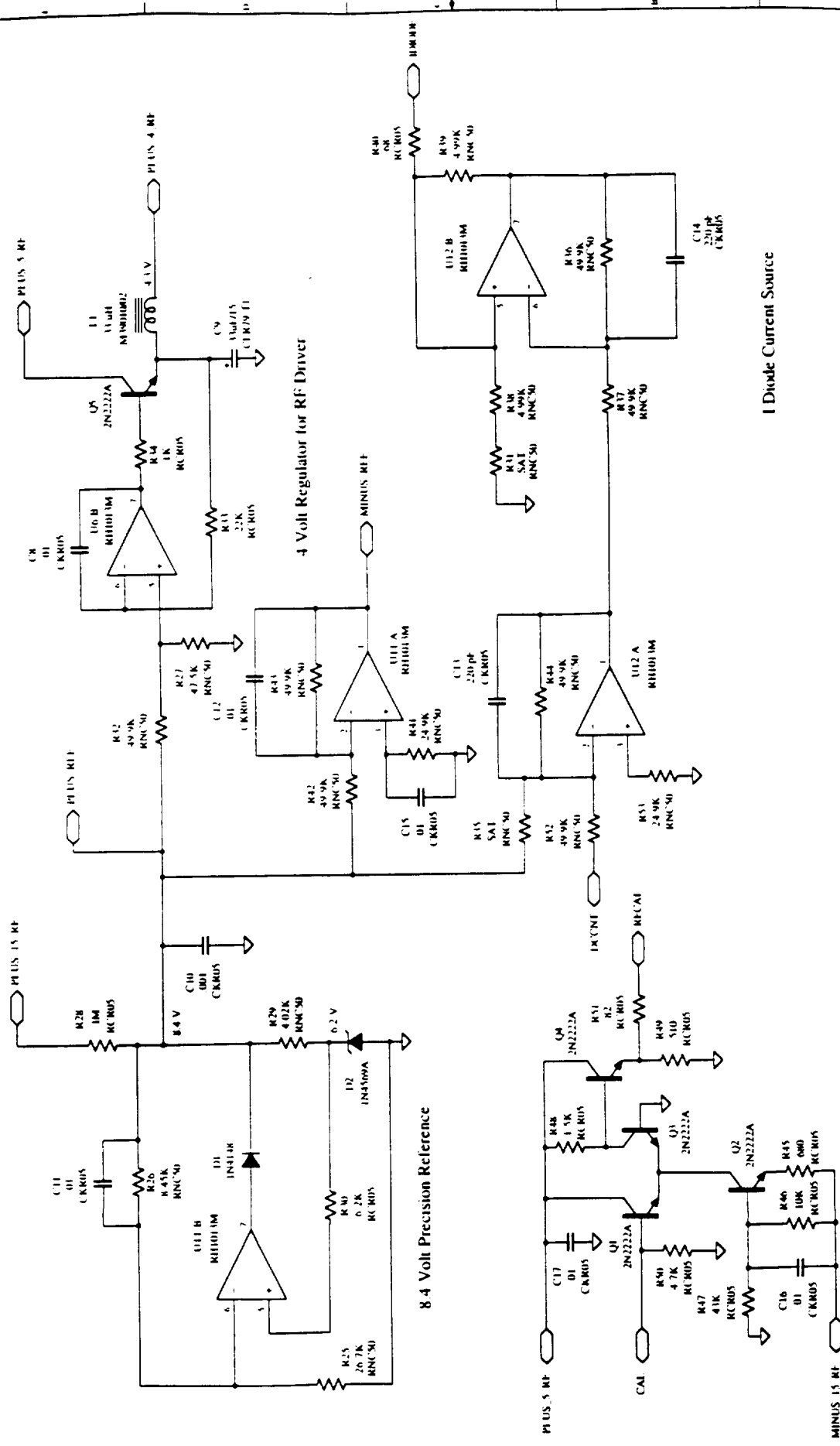


High Voltage Amplifiers

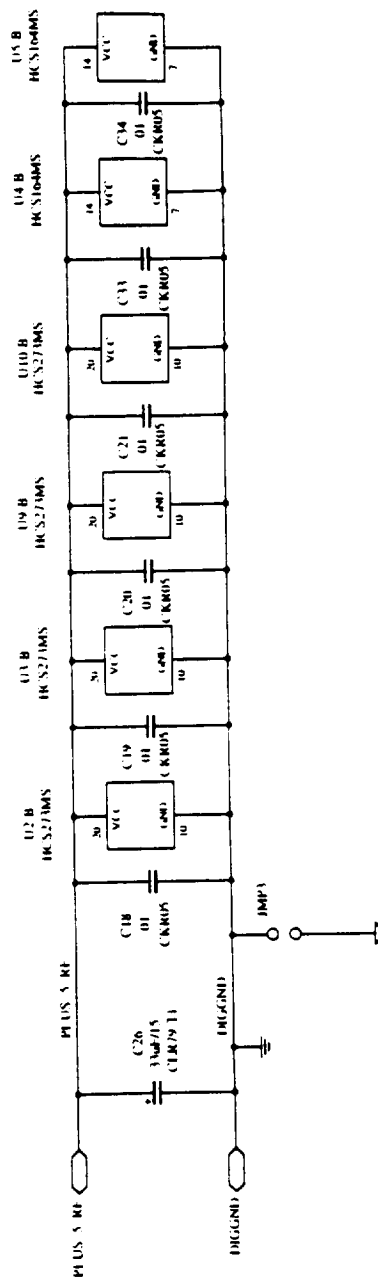
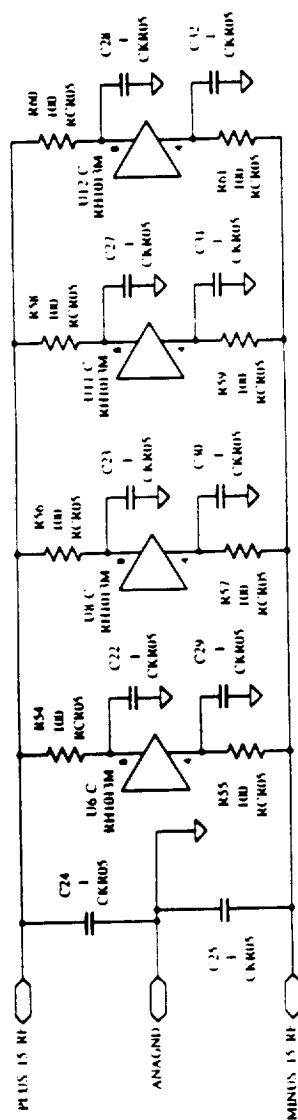
DC Control Voltage Amplifiers

DC Latches and DAC

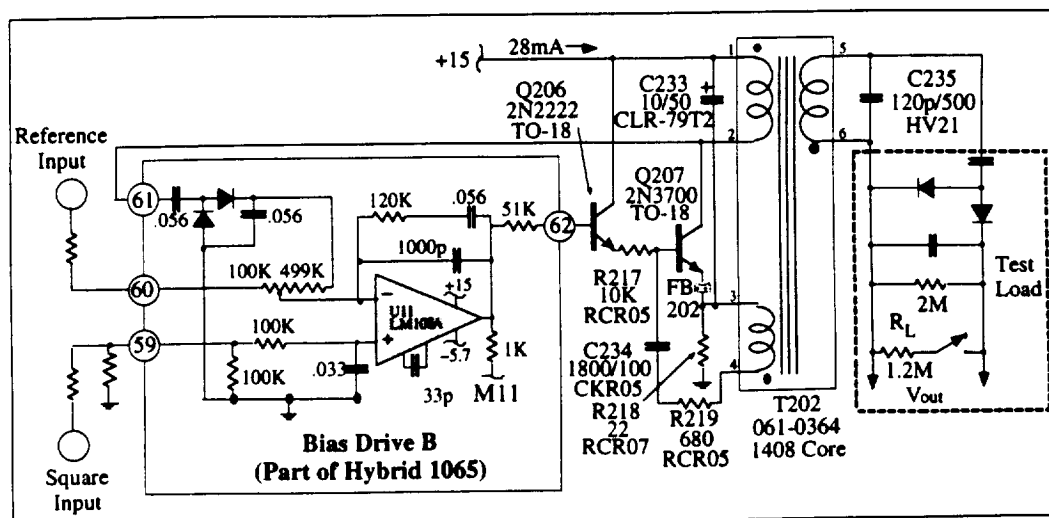
Control Board, Pg. 3	Space Physics Research Laboratory College of Engineering University of Michigan
RF/DC Supply Planet-B NMS	Planet-B - NMS
10355 2-RF/DC Ctrl. Pg. 3	07TK63
061-0355	10355 2-RF/DC Ctrl. Pg. 3
A	061-0355



RF Calibration Driver

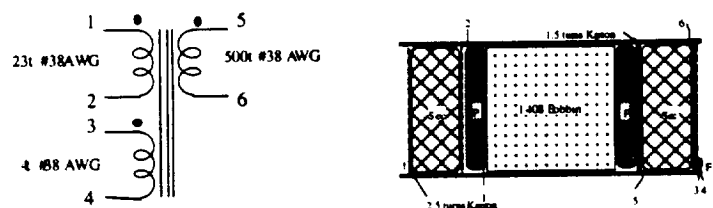


To verify the stability and circuit values the circuit below was used for test:



Test Circuit

The hybrid parts of the circuit were duplicated with discrete parts on the Vector Board. Two cores were wound for the tests. The final core design was wound as shown below. The following transformer specifications were copied from SPRL Dwg. 061-0364, Rev.-, Pg. 5.23.



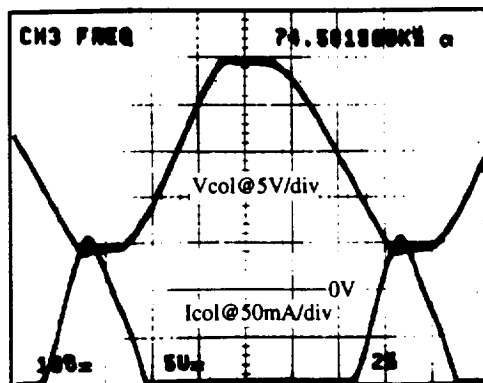
The placement of the feedback winding outside of the resonant secondary is very important to the stability of the resonant circuit. Not also that the AC low side of the secondary winding is next to the primary winding. The feedback winding covers such a small area of the resonant secondary that the increase in capacitance is very small.

To reduce the flux density and hence losses, the volts/turn was reduced from approximately 1 volt/turn to .6 volt/turn. To accommodate the additional turns the wire size was reduced from #30 to #38 on the primary. The secondary resonant current is approximately 20 mA peak. To keep the resonant frequency approximately 75 KHz, the core gap was changed from an A_L 250 core to an A_L 160 core. The resonant frequency is now approximately 77 KHz with a 120 pF resonating capacitor on the secondary.

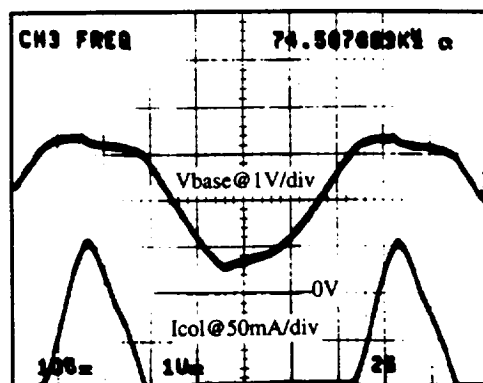
Not shown in the above picture is that the start and finish of the primary and secondary windings come from the same side of the core. i.e. There are no half turns that can cause a PCB current loop that could cause radiated noise.

The following pictures were taken with the circuit values shown on the preceding page. The collector and base voltage pictures were taken with the parallel resistor load. The is the nominal high load expected for two electron multipliers that have a 70 Meg load at 3000 volts.

Not the significant flattening of the curve because of the symmetrical rectifier load that clamps the peak. The sine shape could be improved by the use of a larger resonant capacitor. This would increase the overall power because of the resonant current in the #38 secondary wire.



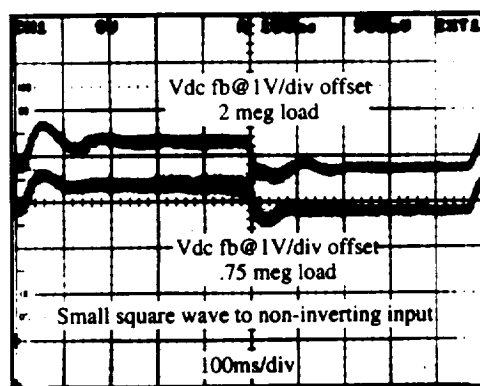
Picture 1



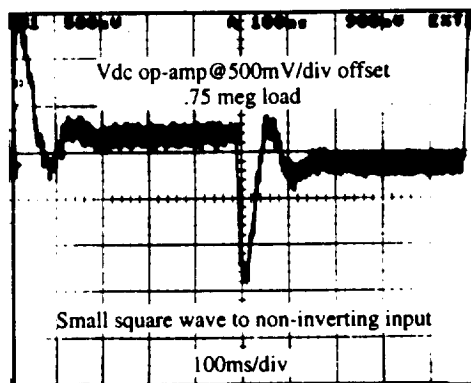
Picture 2

The following two pictures were taken to show the overall loop stability. This stability is significantly influenced by the magnetic design of the transformer. Picture 3 is a double exposure showing the rectified DC feedback voltage (inside the hybrid). This illustrates the difference of changing the load from 2 Meg to .75 Meg. The lighter load has more overshoot and tendency for ringing. At no load the circuit will squeg (oscillate). A 10 Meg DVM load will stabilize the circuit.

Picture 4 is the op-amp voltage with the .75 Meg load. This confirms that the op-amp tracks the DC feedback voltage. This picture can be duplicated on a finished PCB when the hybrid is used.



Picture 3



Picture 4

Summary

This circuit is stable at all reasonable loads. Without a load, circuit oscillations may be observed but will not cause any damage. No temperature tests were run since a flight type of voltage multiplier was not used in the circuit.

061-3054

Batttel Engineering

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September 12, 1995
BE/041/95/330

TELEFAX

To: Bruce Block, SPRL cc. J. Maurer

Fax #: 313-783-5567

From: S. J. Batttel

Page(s) Including Cover: 3

Subject: Lambda Advanced Analog Meeting Report

Message: Hi Bruce- I met with Mike Sullivan and his team at Lambda Advanced Analog (LAA) on Wednesday 9/6/95. The meeting was well supported and, overall, I believe that we came up with a viable plan for both the NRE activity and the flight build.

The primary goal of the meeting was to discuss potential design changes and develop both a plan and timeline for the NRE effort. The recommended plan is as follows:

1. No change will be made to the design to accommodate the 300 kHz switching frequency since the existing design is currently specified up to that value. Based on discussions with the LAA engineers, the frequency design margin is at least 10% with the main degradation factor above that limit being a gradual decrease in efficiency.
2. The overvoltage protection (OVP) will be removed by not installing the parts on the hybrid substrate. This will eliminate the potential OVP failure mode and open up a little room on the substrate to accommodate any size changes (not expected) to the magnetics.
3. The undervoltage lockout (UVL) circuit and auxiliary bias supply will be modified through part changes to support operation down to 15.00 volts (or at least 15.50 volts).
4. The transformer and output inductor turns ratio will be modified as and/or optimized to support operation down to 15.00 volts (or at least 15.50 volts) under our expected load and thermal conditions. Since changing the turns ratio results in discrete voltage steps, it was agreed that a slightly higher voltage on the +/-15 output (such as +/-15.25 volts) was preferable to a slightly lower one.

Please Call (602) 991-9747 if you do not
receive all sheets

5. LAA will provide an SCD outline and support our preparation of a procurement specification with the goal of having a released specification in time for a flight order on 11/1.

6. As part of the NRE activity, LAA will produce a non-flight prototype (meeting form, fit and function) that will be furnished to SPRL for design validation in the 1/1/96 timeframe. Upon completion of testing, it was agreed that the unit would be temporarily returned to LAA for validation of test setups and test software.

In addition to the NRE activity schedule, QA options and cost were discussed.

1. The schedule for all activities was tentatively agreed to be as follows:

- a) 10/1 Contract in place; Start NRE activity.
- b) 11/1 SCD ready; Place flight order to allow procurement of long lead parts.
- b) 12/1 Hold internal review meeting at LAA.
- c) 1/1 Flight unit documentation changes complete; Deliver prototype unit to SPRL; Initiate flight manufacturing.
- d) 2/1 Return prototype unit to LAA.
- e) 4/1 Deliver flight units.

2. From a QA perspective, LAA builds all of their flight units to MIL-H-38534 Class K. This include element evaluation and can have local government QCI (Mike promised to get back to me regarding the specific agency) or GSFC inspection as required. We will need to discuss this further, but it is my belief that we are better off without government inspection. It may prove useful, however, to have Bob Savage of GSFC perform a pre-cap visual.

Qualification will be based on the 87 units already built and tested for INTELSAT since we agreed that the cost and schedule impact of a separate qualification effort could not be accommodated. Given this assumption, we worked hard to minimize the scope of the NRE activity such that the unit basic design, part choices and substate layout would not be changed.

3. The cost of the flight units will be \$18,500 each in quantities of 1 to 4 or \$14,400 each for quantities of 5 or greater. Mike called me yesterday with a ROM cost of \$20,000 for the NRE activity.

NOV-19-95 SUN 11:08

BATTEL ENGINEERING

FAX NO. 6029919748

P.01

061-3060

Battel Engineering

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November 19, 1995

BE/041/95/389

TELEFAX

To: Bruce Block, SPRL cc. J. Maurer, K. Arnett

Fax #: 313-763-5567

From: S. J. Battel

Page(s) Including Cover: 8

Subject: Lambda Test Data

Message: Hi Bruce- Here is preliminary data from LAA for our configuration. Performance is better than what we had with the standard unit although not quite as good as we had hoped. A key concern is performance at very low currents on the +/- 15 volt lines. We will want to do some careful testing in this area when we get the prototype unit.

Dick says that they will be sealing up the proto unit next week and then doing some temperature testing. Dick will also take a crack at redlining the SCD to match what they have observed in the tests. They will be ready for a review on 12/14 as planned.

Best Regards, Steve

FAX COVER SHEET

LAUREN ADVANCED ANALOG INC. 

2270 Martin Ave. Santa Clara, CA 95050 Ph: 408 988-4930 FAX: 408 988-2702

To: Steven J. Battel
Company: Battel Engineering
Phone: 602-991-9747
Fax: 602-991-9748

From: Dick Miller

Date: 18-Nov-95
Pages: 7 including this cover

Subject: Performance Curves for Type 8888 DC/DC Converter

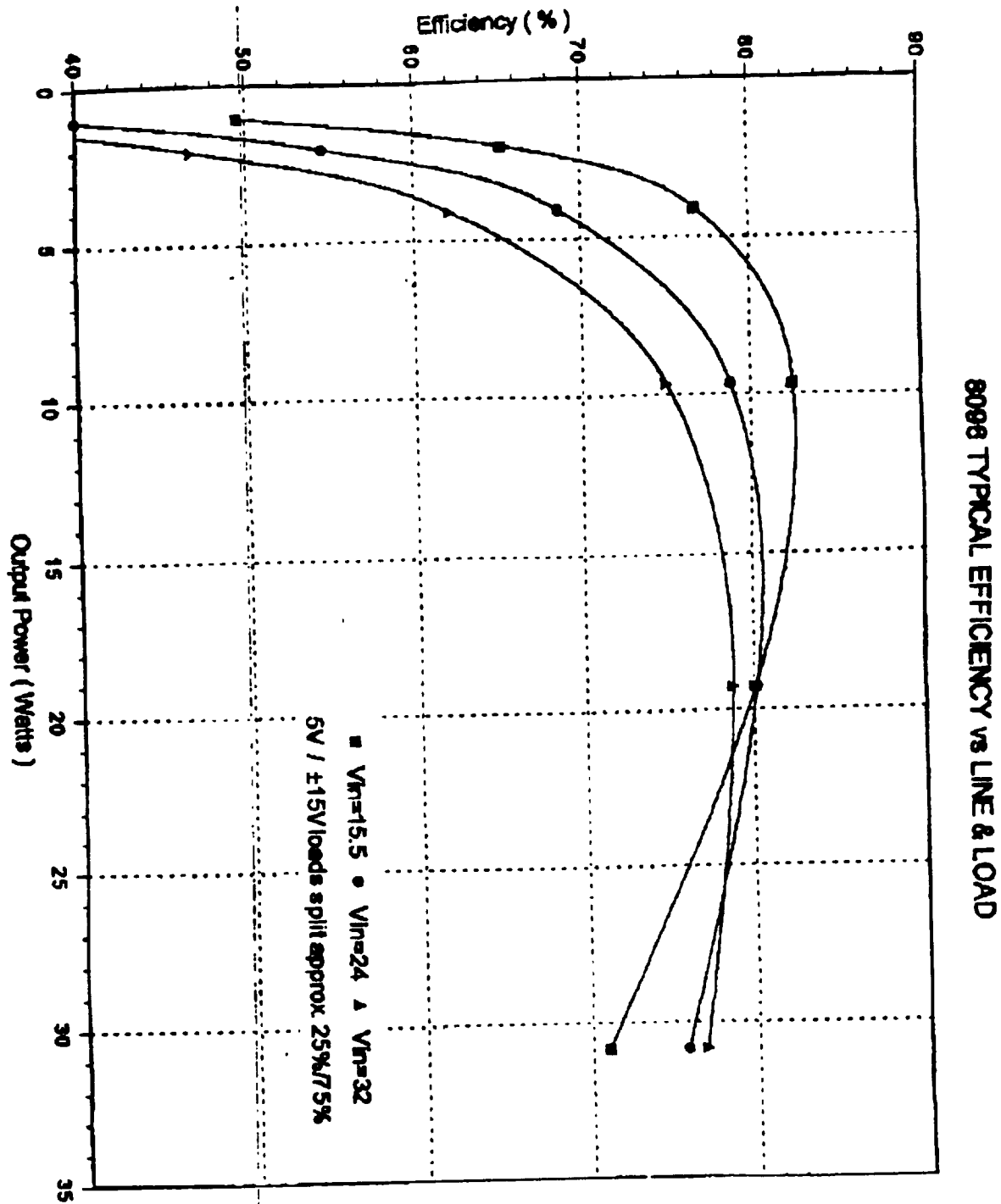
The following pages include a collection of performance plots generated from selected data accumulated by Steve Baker. These data (which include data points outside the specified operating range) will give you an idea about the general performance of the device. We believe it will do the job as intended.

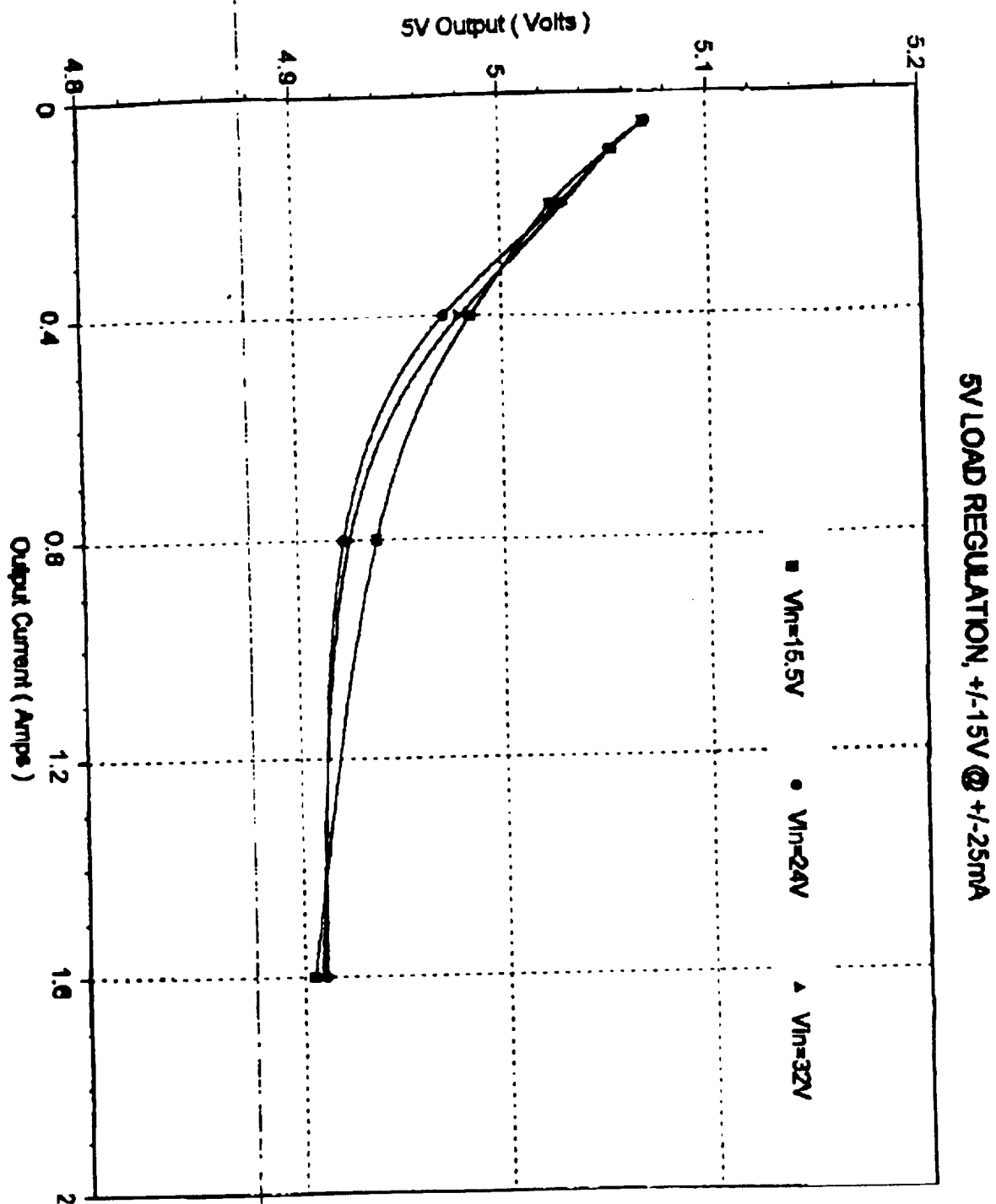
However, because of limitations in the way the test limits were specified in the SCD, we would like to redefine the performance specifications to more closely reflect your requirements.

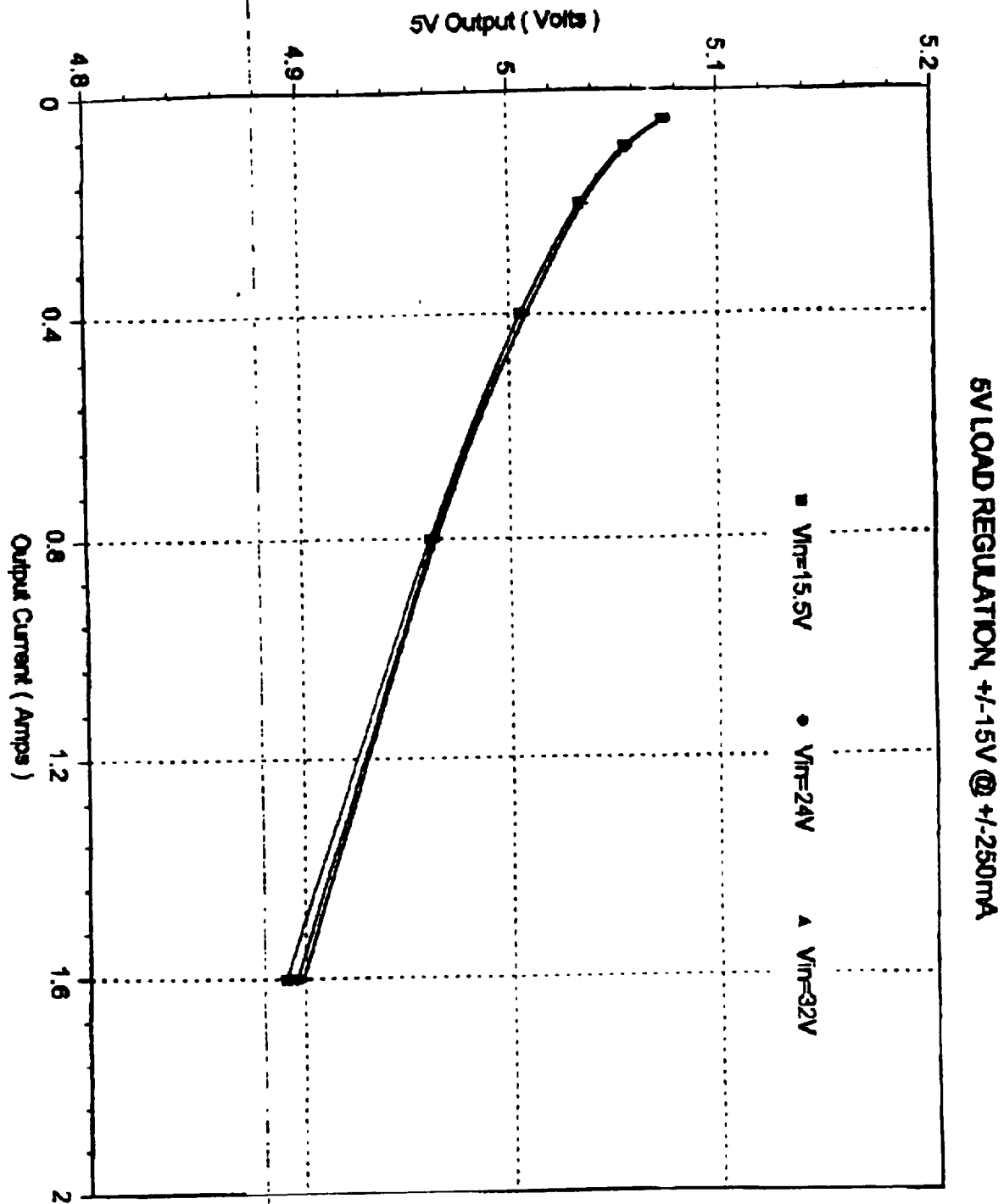
Inspection of the curves shows that as output loading gets very light, regulation rapidly deteriorates and some of the limits in the existing test table cannot be met over all the conditions specified. We do believe that the performance is consonant with your requirement and look for your concurrence on a redefinition.

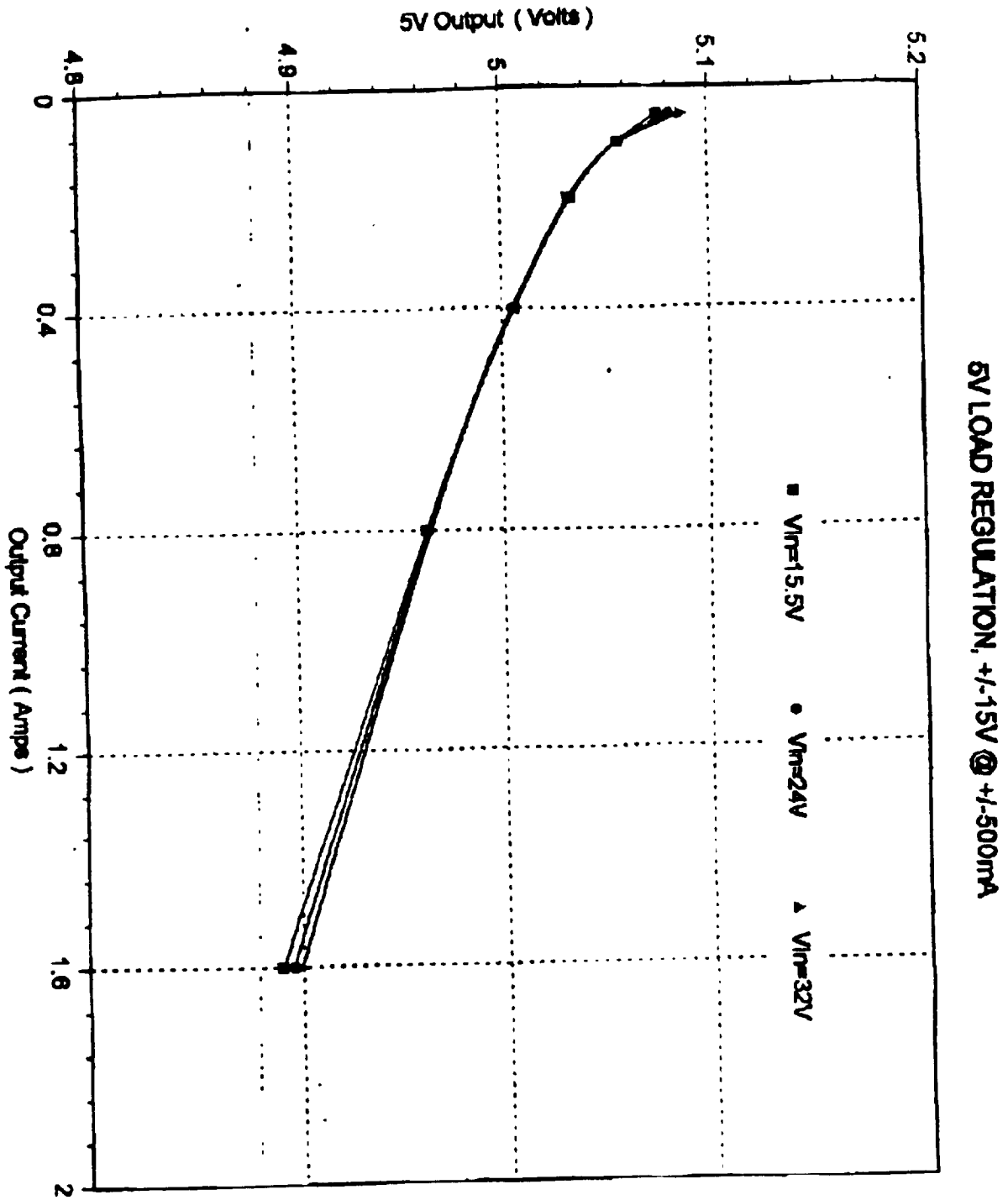
We'll talk after lunch.

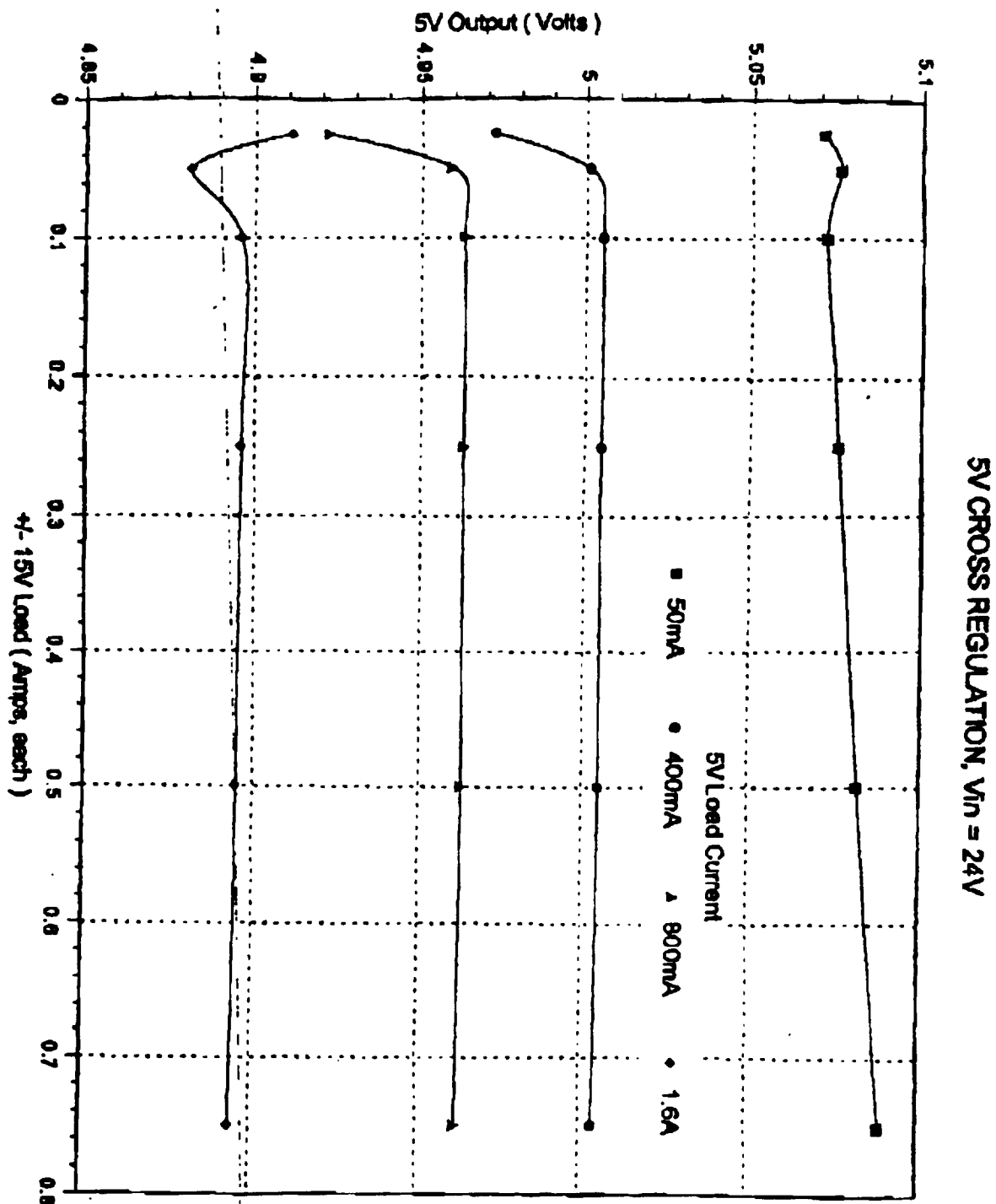
Dick



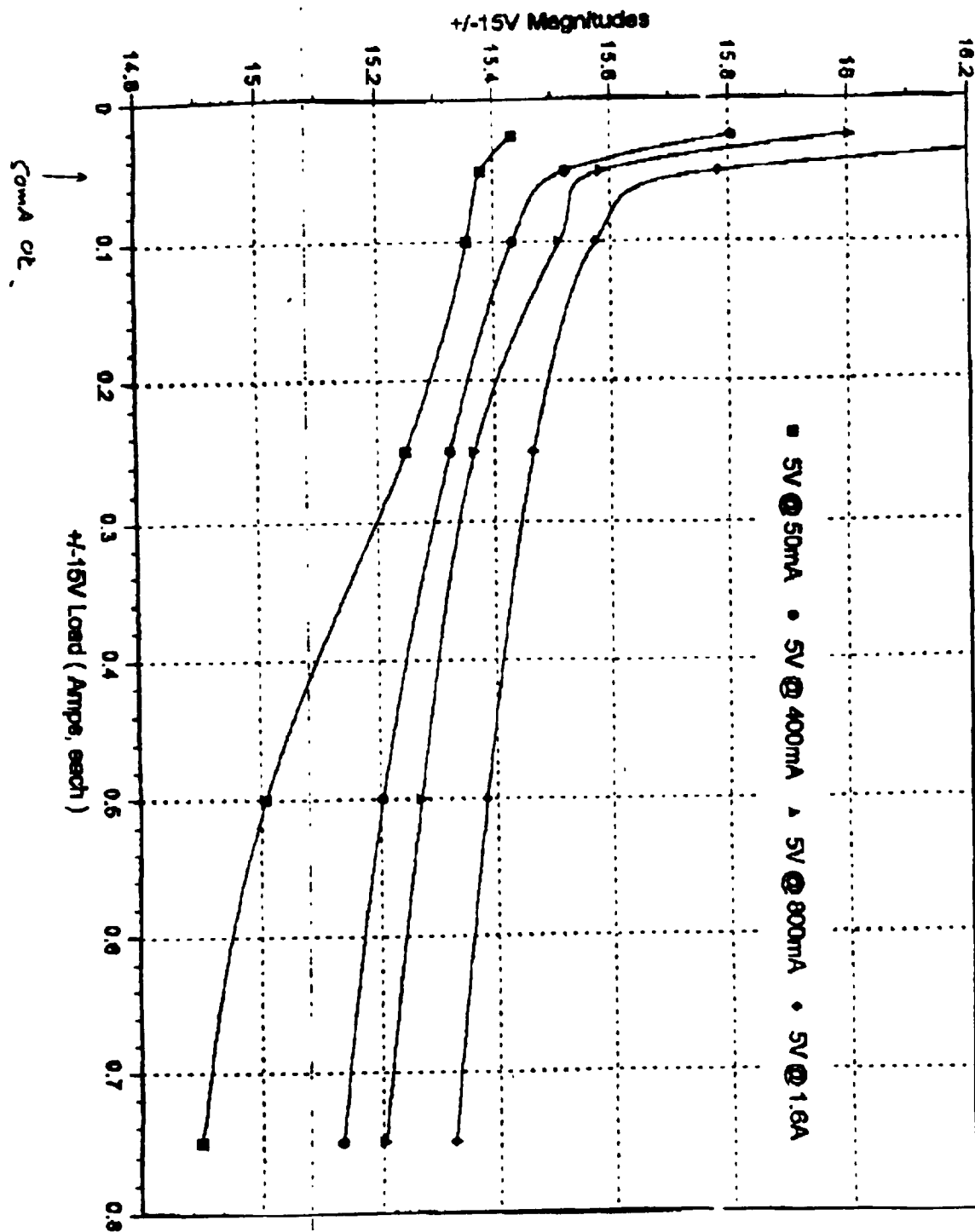


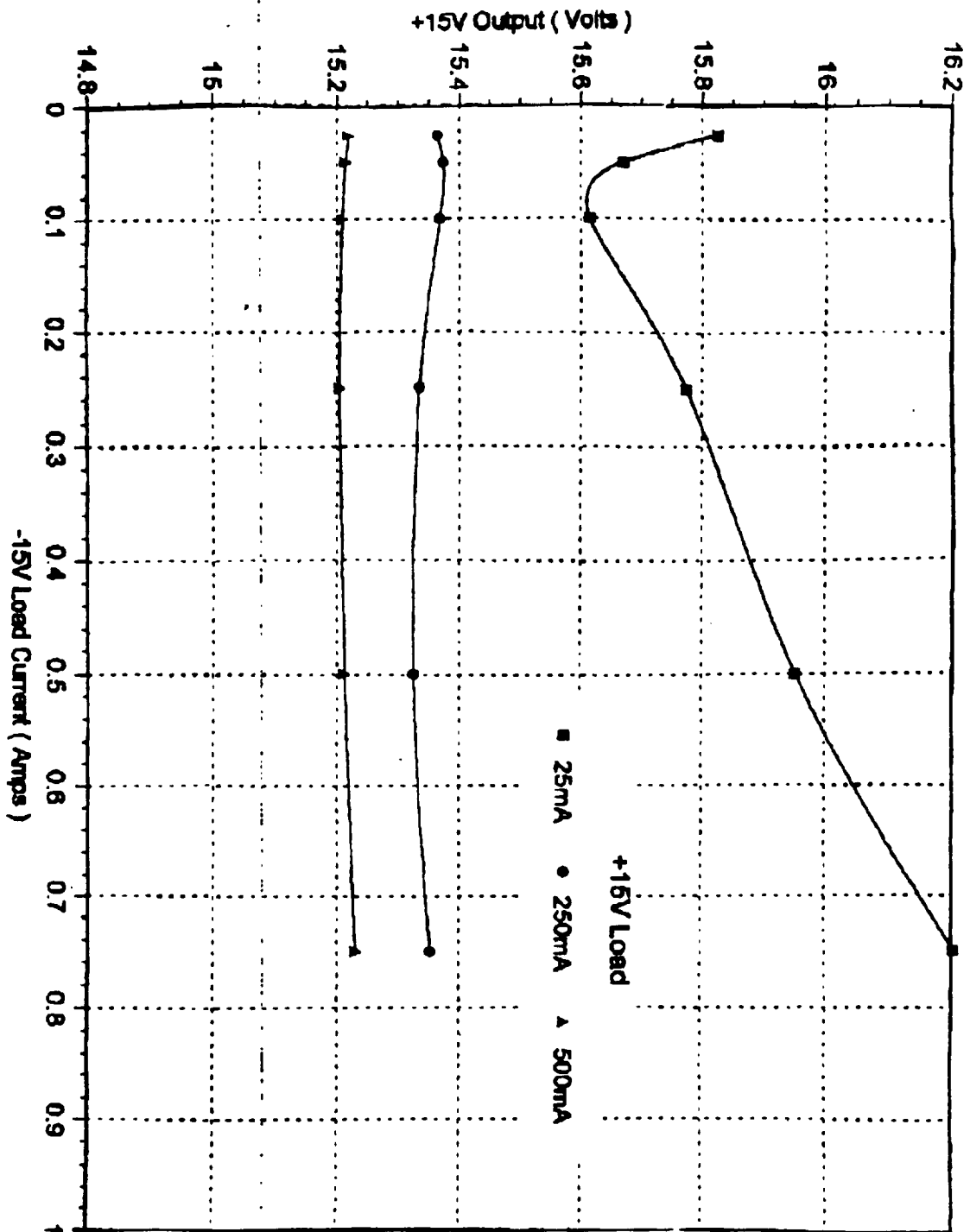






+/-15V LOAD REGULATION, $V_{in} = 24V$





+15V Cross Regulation vs +/-15V Load
5V @ 400mA, Vin = 24V

061-3062

REVISION RECORD

REV LETTER	DESCRIPTION	APPROVED	DATE
A	ECO #12446, Initial Release	<i>RM</i>	22 Sep 95

Type 8096

Radiation Hardened

DC/DC Converter

Rev																		
Sheet																		
Revision Status Of Sheets	Rev	A	A	A	A	A	A	A	A	A	A	A	A	A				
	Sheet	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15		

LAMBDA ADVANCED ANALOG INC. 2270 Martin Avenue, Santa Clara CA 95050
408 988-4930 • FAX 408 988-2702

SOURCE CONTROL DRAWING	PREPARED BY <i>R. Miller</i>		TITLE: Three Output, +5V, ±15.4V DC/DC Converter Hybrid, Radiation Hardened			
	PURCH					
	APPROVED BY					
	DATE OF APPROVAL		DRAWING NO. SCD 171558		REVISION LETTER A	
	SIZE A		CAGE CODE 52467		SHEET 1 OF 12	

DC/DC Converter, Hybrid
Three Output, +5Vdc, ± 15.7 Vdc
Radiation Hardened

1. Scope

- 1.1 Statement of scope. This specification covers the detail requirements for hybrid dc/dc converter microcircuits to be used for space applications.

2. Applicable Documents

- 2.1 Government documents. The following documents form a part of this specification to the extent specified herein. The revision in effect at the time of purchase order placement shall be the minimum revision acceptable.

Specifications

Military

MIL-H-38534	General Specifications for Hybrid Microcircuits
MIL-STD-883	Test Methods and Procedure for Microelectronics
MIL-STD-1686	Electrostatic Discharge Control Program for Protection of Electrical and Electronic Parts, Assemblies and Equipment (Excluding Electrically Initiated Explosive Devices)
MIL-STD-1772	Certification Requirements for Hybrid Microcircuit Facilities and Lines

3. Requirements

- 3.1 Item requirements. Microcircuits supplied to this drawing shall be in accordance with 1.2.2 of MIL-STD-883, "Provisions for the use of MIL-STD-883 in conjunction with non-compliant non-JAN devices" and shall meet all of the MIL-H-38534 class K requirements with the exceptions and additions specified herein. The quality assurance requirements shall be in accordance with MIL-H-38534 class K option 2 (end-of-line)

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- 3.1.1 Absolute maximum ratings. Stresses above the absolute maximum rating may cause permanent damage to the device. Extended operation at the maximum levels may degrade performance and affect reliability.

Input voltage range	-0.5 Vdc to +50 Vdc
Output power	internally limited to 52 W typical
Lead temperature (soldering, 10 seconds)	+300°C
Operating temperature range (case)	-55°C to +125°C
Storage temperature range (case)	-65°C to +135°C

- 3.2 Design and construction. The design and construction shall be as specified in MIL-H-38534 for class K with the exceptions and additions specified herein.

- 3.2.1 Case outline. The case outline shall be in accordance with figure 1.

- 3.2.2 Terminal connections. The terminal connections shall be as specified in table I.

- 3.2.3 Block diagram. The block diagram shall be as specified in figure 2.

- 3.2.4 Weight. The weight shall not exceed 120 grams.

- 3.2.5 Lead finish. The lead finish shall be finish A in accordance with MIL-H-38534.

- 3.3 Electrical performance characteristics. Unless otherwise specified herein, the electrical performance characteristics are as specified in table II and shall apply over the full specified operating temperature range.

- 3.4 Electrical test requirements. The electrical test requirements shall be the subgroups specified in table IV. The electrical tests for each subgroup are described in table III.

- 3.5 Radiation hardness assurance (RHA). The RHA shall be in accordance MIL-H-39534 level R except with the additional requirements and exceptions specified herein.

- The neutron fluence level shall be 3×10^{12} n/cm².
- The transient ionizing dose shall be $\geq 1 \times 10^{11}$ rad (si)/s for survival.
- Single event effects - The microcircuit shall be immune to single-event upset (SEU), latchup, and permanent damage (including burnout and gate rupture). The microcircuit is considered immune to single event effects if its linear energy transfer (LET) threshold is greater than 80 MeV-cm²/mg.

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Scale: NONE			Sheet 3 of 12

3.6 Marking. Marking shall be in accordance with MIL-H-38534 except as specified herein. The part or identifying number (PIN) shall be as listed in table V and the following marking shall be included on each microcircuit as a minimum.

- a. Index point or pin 1 identification.
- b. PIN.
- c. Lot identification code or date code.
- d. Manufacturer's CAGE code, and logo, mark, name or other distinctive identification.
- e. ESD identifier.
- f. Serial number (see para. 3.6.1).

3.6.1 Serial number. Serial number shall be assigned in sequence and shall not be used more than once.

3.7 Line certification. Microcircuits supplied to this drawing shall be fabricated on a line certified in accordance with MIL-H-38534 to manufacture compliant hybrids.

4. Quality Assurance Provisions

4.1 Quality assurance provisions shall be in accordance with MIL-H-38534 class K option 2 with the exceptions and additions specified herein.

4.2 Screening. Screening shall be in accordance with MIL-H-38534 class K with the exceptions and additions specified herein. Screening shall be conducted on all devices prior to quality conformance inspection.

- a. The minimum nondestructive bond pull limits shall be the value specified in method 2023 of MIL-STD-883.
- b. Burn-in test, method 1015 of MIL-STD-883.
 - 1. Test condition C, steady state. The test circuit shall be maintained by the manufacturer under document revision level control and shall be made available upon request. Also, the test circuit shall specify the inputs, outputs, biases, and power dissipation, as applicable, in accordance with the intent specified in test method 1015 of MIL-STD-883.
- c. Interim and final electrical test parameters shall be as specified in table III.

4.3 Quality conformance inspection. Quality conformance inspection shall be in accordance with MIL-H-38534 class K except as specified herein.

4.3.1 Group A inspection. Group A inspection shall be in accordance with MIL-H-38534 class K except as follows:

- a. Tests shall be as specified in table IV herein.
- b. Subgroups 7,8,9,10 and 11 shall be omitted.

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Scale: NONE			Sheet 4 of 12

4.3.2 Group B inspection. Group B inspection shall be in accordance with MIL-H-38534 class K.

4.3.3 Group C inspection by similarity When specified in the purchase order, Group C inspection requirement shall be satisfied in its entirety by the application of similarity data from manufacturing lots of parts passing the group C criteria of MIL-H-38534 for class K. Such data shall be made available for inspection when requested.

4.3.4 Group E inspection by similarity. Similarity test data and analysis shall be used to meet the RHA requirements specified herein.

4.4 Data. A data review shall be performed against the following requirements prior to acceptance of the lot. The manufacturer shall create and maintain, as a minimum, with each lot of parts the following:

- a. Completed assembly, screening, and qualification/QCI lot travelers.
- b. Screening electrical test variables data, identified by serial number, for each device at each specified temperature. This includes devices which fail and are accountable in the PDA. PDA calculations or results are to be recorded on the traveler.
- c. Group A variables data, identified by serial number.
- d. Group B variables data, identified by serial number.
- e. Catastrophic failure data and failure analysis results, if applicable.

5. Packaging

5.1 Packaging requirements. The requirements for packaging shall be in accordance with MIL-H-38534.

6.0 Notes

6.1 Intended use. Microcircuits conforming to this drawing are intended for use in radiation hard space level microcircuit applications (original equipment), design applications, and logistics purposes.

6.2 Approved sources of supply. The approved source(s) of supply shall be as shown table V.

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Table I
Terminal Connections

Terminal Number	Terminal Symbol
1	+ V input
2	Input return
3	ENABLE
4	Sync
5	No connection
8	No connection
9	- 15 Vdc output
10	15 Vdc output return
11	+ 15 Vdc output
12	Chassis
13	+ 5 Vdc output
14	5 Vdc output return

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Table II
Electrical Performance Characteristics

Parameter	Symbol	Conditions unless otherwise specified $-55^{\circ}\text{C} \leq T_C \leq +125^{\circ}\text{C}$, $V_{IN} = 24\text{Vdc} \pm 5\%$, $C_L = 0$	Limit MIN	Limit MAX	Units
Output voltage accuracy	V_{OUT}	$I_{OUT} = 350\text{mA}$ dc $T_C = +25^{\circ}\text{C}$ (main) $I_{OUT} = 250\text{mA}$ dc $T_C = +25^{\circ}\text{C}$ (dual)	4.95 ± 15.00	5.05 ± 15.63	Vdc
Output power Note 3	P_{OUT}	$15.5\text{Vdc} < V_{IN} < 31\text{Vdc}$	2	20	W
Output current Note 3	I_{OUT}	$15.5\text{Vdc} < V_{IN} < 31\text{Vdc}$ (main) (dual)	80 50	800 500	mA
Line regulation Note 1	VR_{LINE}	$175\text{mA} < I_{OUT} < 800\text{mA}$ dc (main) $15.5\text{Vdc} < V_{IN} < 31\text{Vdc}$ $\pm 50\text{mA} < I_{OUT} < \pm 500\text{mA}$ dc (dual)	-15 -60	+15 +60	mV
Load regulation Note 2	VR_{LOAD}	$175\text{mA} < I_{OUT} < 800\text{mA}$ dc (main) $15.5\text{Vdc} < V_{IN} < 31\text{Vdc}$ $\pm 50\text{mA} < I_{OUT} < \pm 500\text{mA}$ dc (dual)	-100 -250	+100 +250	mV
Cross regulation Note 6	VR_{CROSS}	$15.5\text{Vdc} < V_{IN} < 31\text{Vdc}$ (main) (dual)	-10 -250	+10 +250	mV
Total regulation	VR	All conditions of Line, Load, Cross Regulation, Aging, Temperature and Radiation (main) (dual)	4.85 14.5	5.15 16.5	V
Input current	I_{IN}	$I_{OUT} = \text{minimum rated}$, Pin 3 open Pin 3 shorted to pin 2 (disabled)		250 8	mA
Output ripple voltage Note 4	V_{RIP}	$I_{OUT} = 800\text{mA}$ dc (main), $\pm 500\text{mA}$ dc (dual)		100	mV _{p,p}
Input ripple current Note 4	I_{RIP}	$I_{OUT} = 800\text{mA}$ dc (main), $\pm 500\text{mA}$ dc (dual)		100	mA _{p,p}
Switching frequency	F_S	Synchronization input open. (pin 6)	225	275	kHz
Efficiency	Eff	$I_{OUT} = 800\text{mA}$ dc (main), $\pm 500\text{mA}$ dc (dual)	75		%
Enable Input open circuit voltage drive current (sink) voltage range			3.0 -0.5	5.0 100 50	V μA V
Synchronization Input frequency range pulse high level pulse low level pulse rise time pulse duty cycle		External clock signal on Sync. input (pin 4)	225 4.5 -0.5 40 20	310 10.0 0.25 V 80	Khz V V V/ μS %
Power dissipation, load fault	P_D	Short circuit, any output		7.5	W

Size A	Cage Code. 52467	Drawing No. SCD 171558	Rev A
Scale: NONE			Sheet 7 of 12

Table II
Electrical Performance Characteristics
(Continued)

Parameter	Symbol	Conditions unless otherwise specified $-55^{\circ}\text{C} \leq T_c \leq +125^{\circ}\text{C}$, $V_{IN}=24\text{Vdc} \pm 5\%$, $C_L=0$	Limit MIN	Limit MAX	Units
Output response to step load changes <i>Notes 5, 9</i>	V_{TLD}	10% Load to/from 50% load 50% Load to/from 100% load	-200 -200	200 200	mV _{PK}
Recovery time from step load changes <i>Notes 9, 10</i>	T_{TLD}	10% Load to/from 50% load 50% Load to/from 100% load		200 200	μS
Output response to step line changes <i>Notes 8, 9</i>	V_{TLN}	$I_{OUT} = 800 \text{ mAdc}$ (main) $V_{IN} = 15.5\text{V to/from } 31 \text{ V}$ $I_{OUT} = \pm 500 \text{ mAdc}$ (dual)	-350 -1050	350 1050	mV _{PK}
Recovery time from step line changes <i>Notes 8, 9, 11</i>	T_{TLN}	$I_{OUT} = 800 \text{ mAdc}$ (main) $V_{IN} = 15.5\text{V to/from } 31 \text{ V}$ $I_{OUT} = \pm 500 \text{ mAdc}$ (dual)		500 500	μS
Turn on overshoot	V_{OS}	(main) $I_{OUT} = \text{minimum and full rated}$ (dual)		500 1500	mV
Turn on delay <i>Note 12</i>	T_{DLY}	$I_{OUT} = \text{minimum and full rated}$	5	20	mS
Capacitive load <i>Notes 7, 8</i>	CL	(main) No effect on DC performance (dual)		500 100	μF
Isolation	ISO	500VDC Input to Output or any pin to case (except pin 12)	100		M Ω

Notes to Table II Electrical Performance Characteristics

1. Parameter measured from 24V to 15.5V or to 31V while loads remain fixed.
2. Parameter measured from nominal to minimum or maximum load conditions while line remains fixed.
3. Parameter verified by dynamic load regulation tests.
4. Guaranteed for a bandwidth of DC to 20Mhz. Tested using a 20Khz to 2Mhz bandwidth.
5. Load current is stepped for output under test while other outputs are fixed at half rated load.
6. Load current is fixed for output under test while other output loads are varied for any combination of minimum to maximum.
7. A capacitive load of any value from 0 to the specified maximum is permitted without compromise to DC performance. A capacitive load in excess of the maximum limit may interfere with the proper operation of the converter's short circuit protection, causing erratic behavior during turn on.
8. Parameter is tested as part of design characterization or after design or process changes. Thereafter, parameters are guaranteed to the limits specified in the table.
9. Load transient rate of change, $dv/dt \leq 2 \text{ A}/\mu\text{Sec}$.
10. Recovery time is measured from the initiation of the transient to where V_{OUT} has returned to within $\pm 1\%$ of its steady state value.
11. Line transient rate of change, $di/dt \leq 50 \text{ V}/\mu\text{Sec}$.
12. Turn on delay time is for either a step application of input power or a logical low to high transition on the inhibit pin (pin 3) while power is present at the input.

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**Table III
Group A Electrical Tests**

Test	Symbol	Conditions unless otherwise specified $-55^{\circ}\text{C} \leq T_C \leq +125^{\circ}\text{C}$, $V_{IN} \pm 24\text{Vdc} \pm 5\%$, $C_L = 0$	Group A Subgroups	Limit MIN	Limit MAX	Units
Output voltage accuracy	V_{OUT}	$I_{OUT} = 350\text{mA dc}$ (main) $I_{OUT} = \pm 250\text{mA dc}$ (dual)	1, 2, 3 1, 2, 3	4.95 ± 15.00	5.05 ± 15.63	V
Output power <i>Note 1</i>	P_{OUT}	$V_{IN} = 15.5\text{V}, 24\text{V}, 31\text{V}$	1, 2, 3	2	20	W
Output current <i>Note 1</i>	I_{OUT}	$V_{IN} = 15.5\text{V}, 24\text{V}, 31\text{V}$ (main) (dual)	1, 2, 3 1, 2, 3	80 50	800 500	mA
Output regulation <i>Note 4</i>	VR	$I_{OUT} = 175, 350, 800\text{mA dc}$ (main) $V_{IN} = 15.5\text{V}, 24\text{V}, 31\text{V}$ $I_{OUT} = \pm 50, \pm 250, \pm 500\text{mA dc}$ (dual)	1, 2, 3 1, 2, 3	4.85 14.5	5.15 16.5	V
Input current	I_{IN}	$I_{OUT} = \text{minimum rated, Pin 3 open}$ Pin 3 shorted to pin 2 (disabled)	1, 2, 3 1, 2, 3		250 8	mA
Output ripple <i>Note 2</i>	V_{RIP}	$I_{OUT} = 800\text{mA main, } \pm 500\text{mA dual}$	1, 2, 3		100	mV _{p-p}
Input ripple <i>Note 2</i>	I_{RIP}	$I_{OUT} = 800\text{mA main, } \pm 500\text{mA dual}$	1, 2, 3		100	mA _{p-p}
Switching frequency	F_S	Synchronization pin (pin 6) open	4, 5, 6	225	275	KHz
Efficiency	Eff	$I_{OUT} = 800\text{mA main, } \pm 500\text{mA dual}$	1, 2, 3	75		%
Power dissipation, load fault	P_D	Short circuit, any output	1, 2, 3		7.5	W
Output response to step load changes <i>Notes 3, 5</i>	V_{TL}	10% Load to/from 50% load 50% Load to/from 100% load	4, 5, 6 4, 5, 6	-200 -200	200 200	mV _{pk}
Recovery time from step load changes <i>Notes 5, 6</i>	T_{TL}	10% Load to/from 50% load 50% Load to/from 100% load	4, 5, 6 4, 5, 6		200 200	μS
Turn on overshoot	V_{OS}	(main) $I_{OUT} = \text{minimum and full rated}$ (dual)	4, 5, 6 4, 5, 6		500 1500	mV
Turn on delay <i>Note 7</i>	T_{OLY}	$I_{OUT} = \text{minimum and full rated}$	4, 5, 6	5	20	mS
Isolation	ISO	500VDC Input to output or any pin to case (except pin 12)	1	100		M Ω

Notes to Table III Group A Electrical Tests

- Parameter verified during dynamic load regulation tests.
- Guaranteed for DC to 20 MHz bandwidth. Tested using a 20KHz to 2MHz bandwidth.
- Load current is stepped for output under test while other outputs are fixed at half rated load.
- Each output is measured for all combinations of line and load. Only the minimum and maximum readings for each output are recorded.
- Load step transition time $\geq 10\mu\text{S}$.
- Recovery time is measured from the initiation of the transient to where V_{OUT} has returned to within $\pm 1\%$ of its steady state value.
- Turn on delay time is for either a step application of input power or a logical low to high transition on the inhibit pin (pin 3) while power is present at the input.

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Table IV
Electrical Test Requirements.

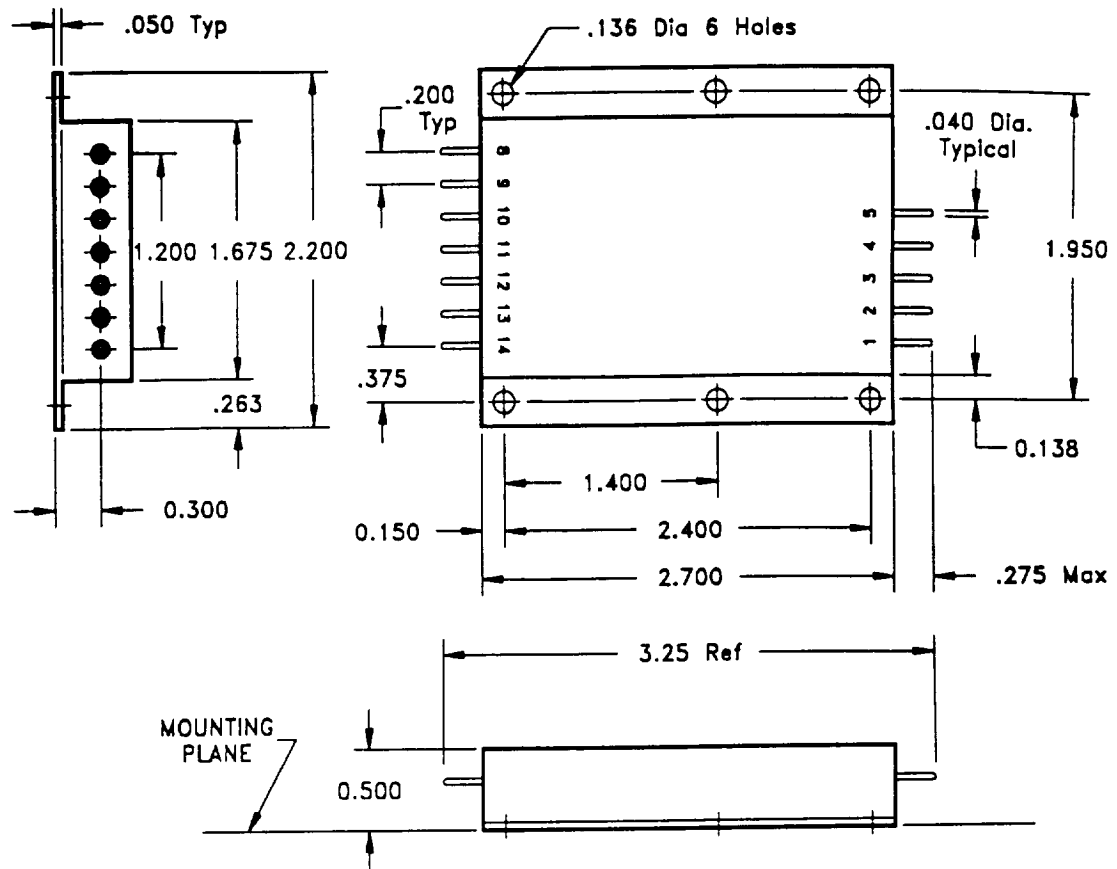
MIL-H-38534 Test Requirements	Subgroups (in accordance with MIL-H-38534, group A test table)
Interim electrical parameters	1 (tests 1, 2, 3, 5, 7 & 8 of table III herein only)
Final electrical test parameters	1* 2, 3, 4, 5, 6
Group A test requirements	1 2, 3, 4, 5, 6
Group C end-point electrical parameters	1, 4
Group E end-point electrical parameters	1, 4

* PDA applies to subgroup 1.

Table V
Approved Sources of Supply

Manufacturer's CAGE code	Manufacturer's name and address
52467	Lambda Advanced Analog 2270 Martin Avenue Santa Clara, CA 95050-2781

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Notes:

Dimensions are in inches.
 Unless otherwise specified, tolerances are

∠	= ± 2°
.XX	= ± .01
.XXX	= ± .005

Base Plate Flatness 0.003 maximum.

Figure 1.
Physical Outline

Size	Cage Code.	Drawing No.	Rev
A	52467	SCD 171558	A
Scale: NONE			Sheet 11 of 12

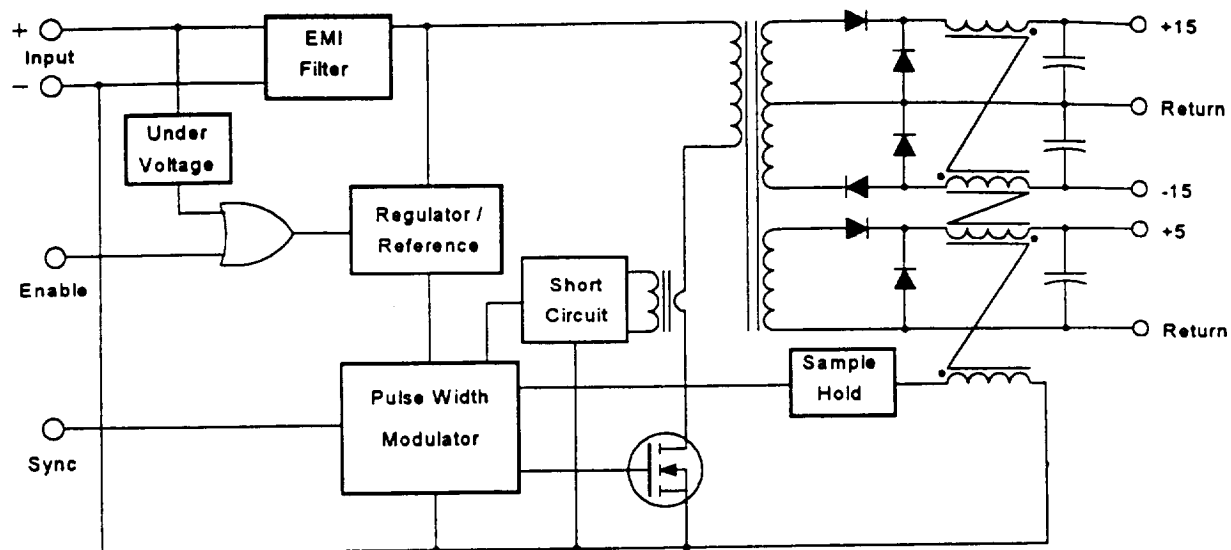
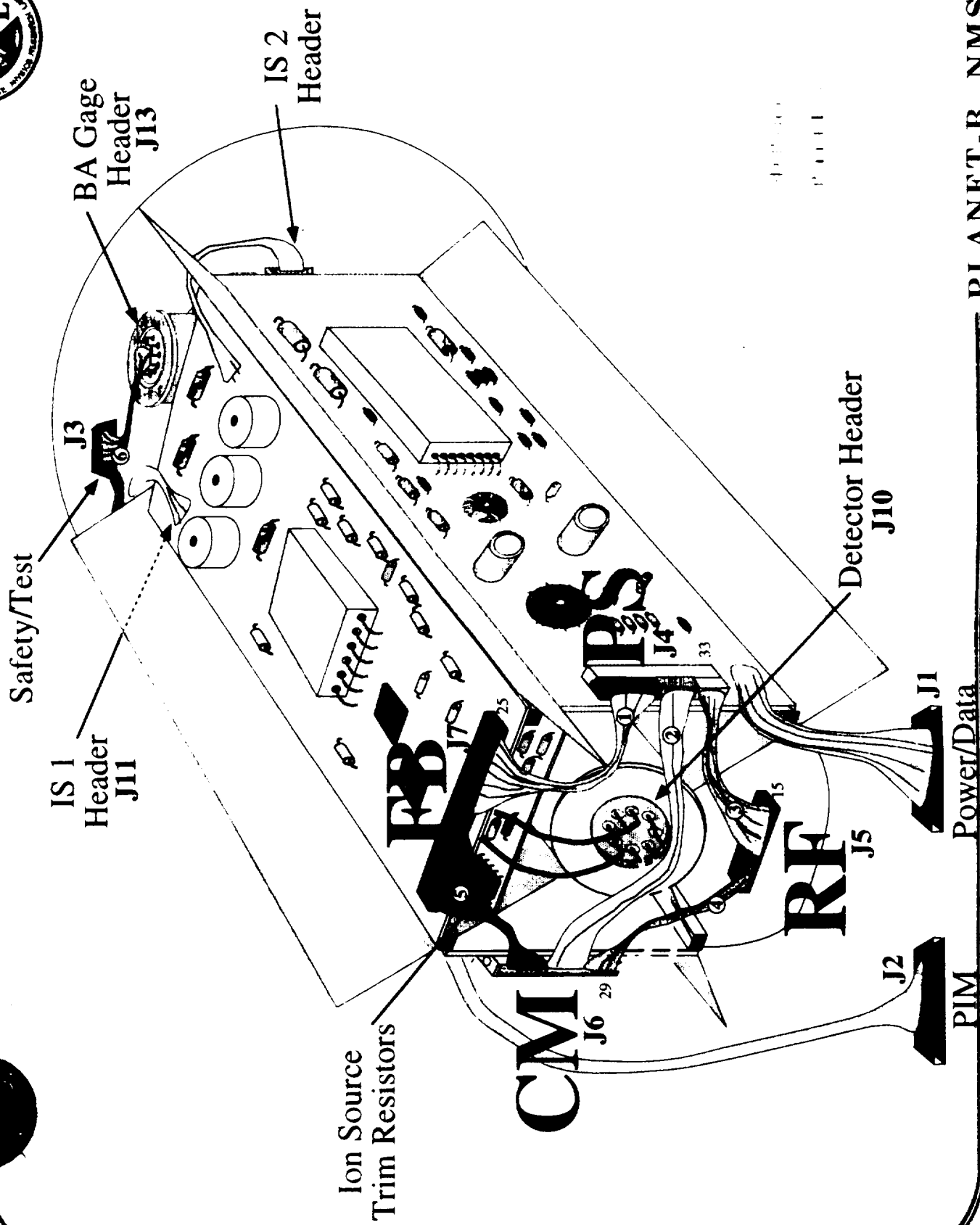
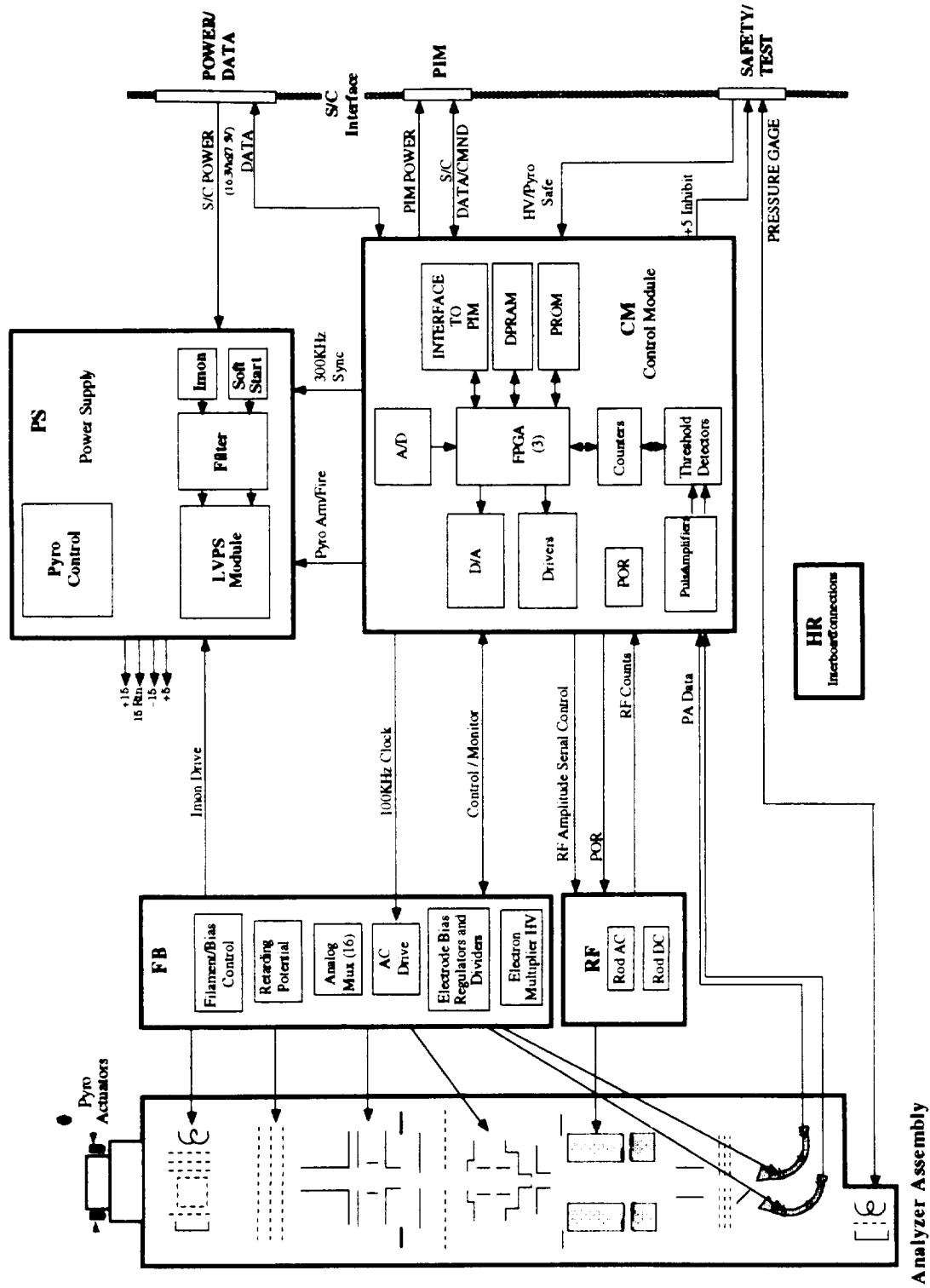
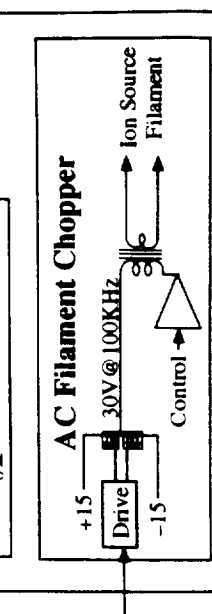
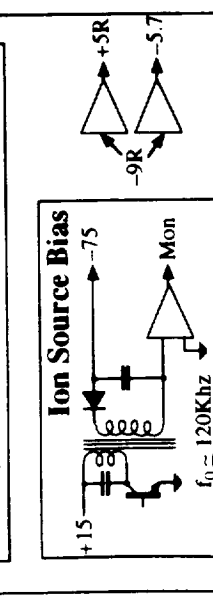
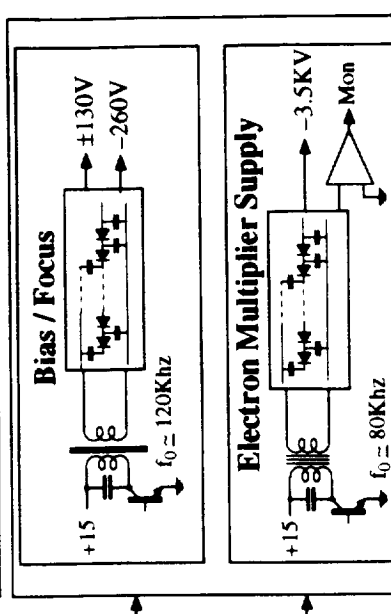
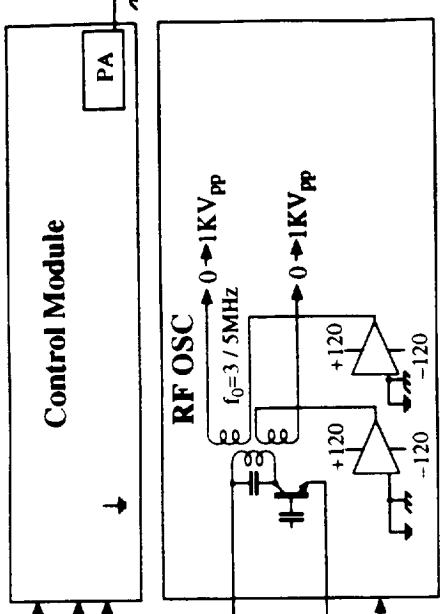
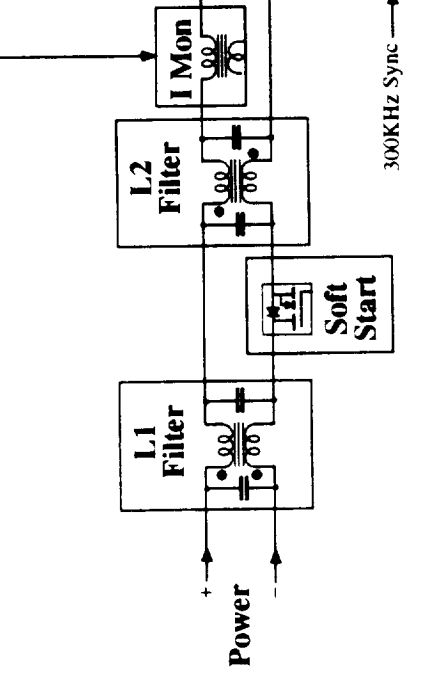
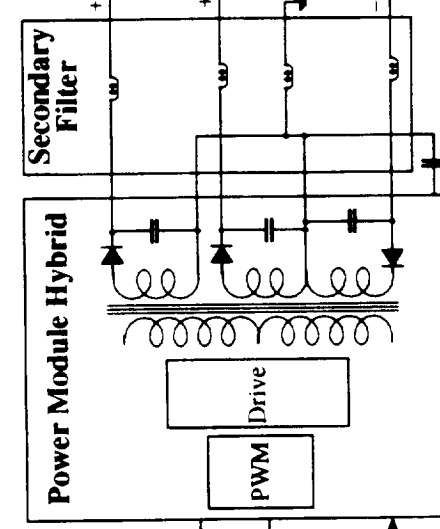
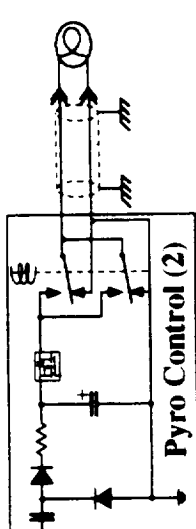


Figure 2.
Block Diagram

Size A	Cage Code. 52467	Drawing No. SCD 171558	Rev A
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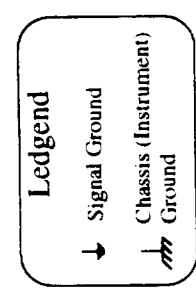




15V @ 100KHz

Power

300KHz Sync



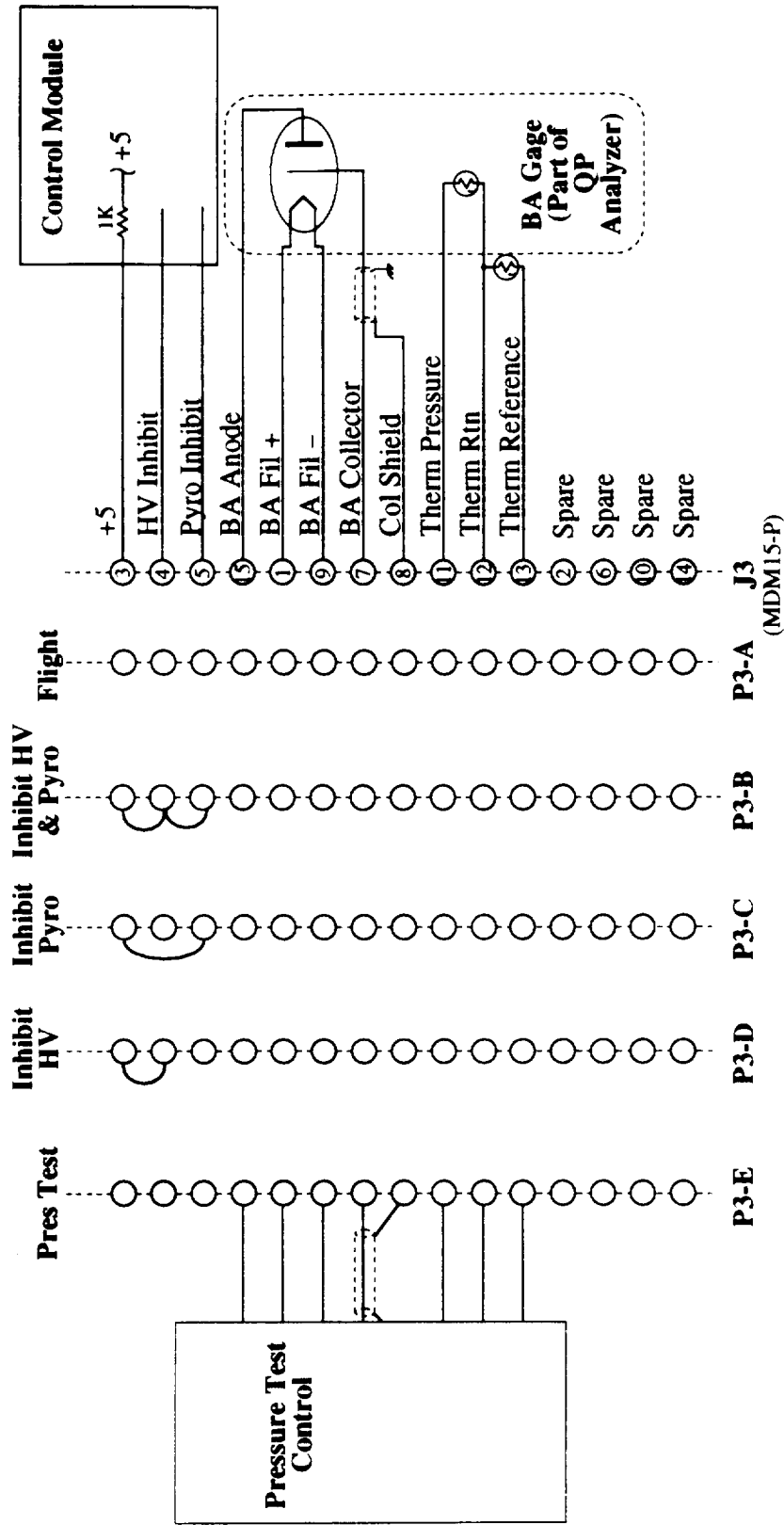
Space Physics Research Laboratory
College of Engineering
University of Michigan

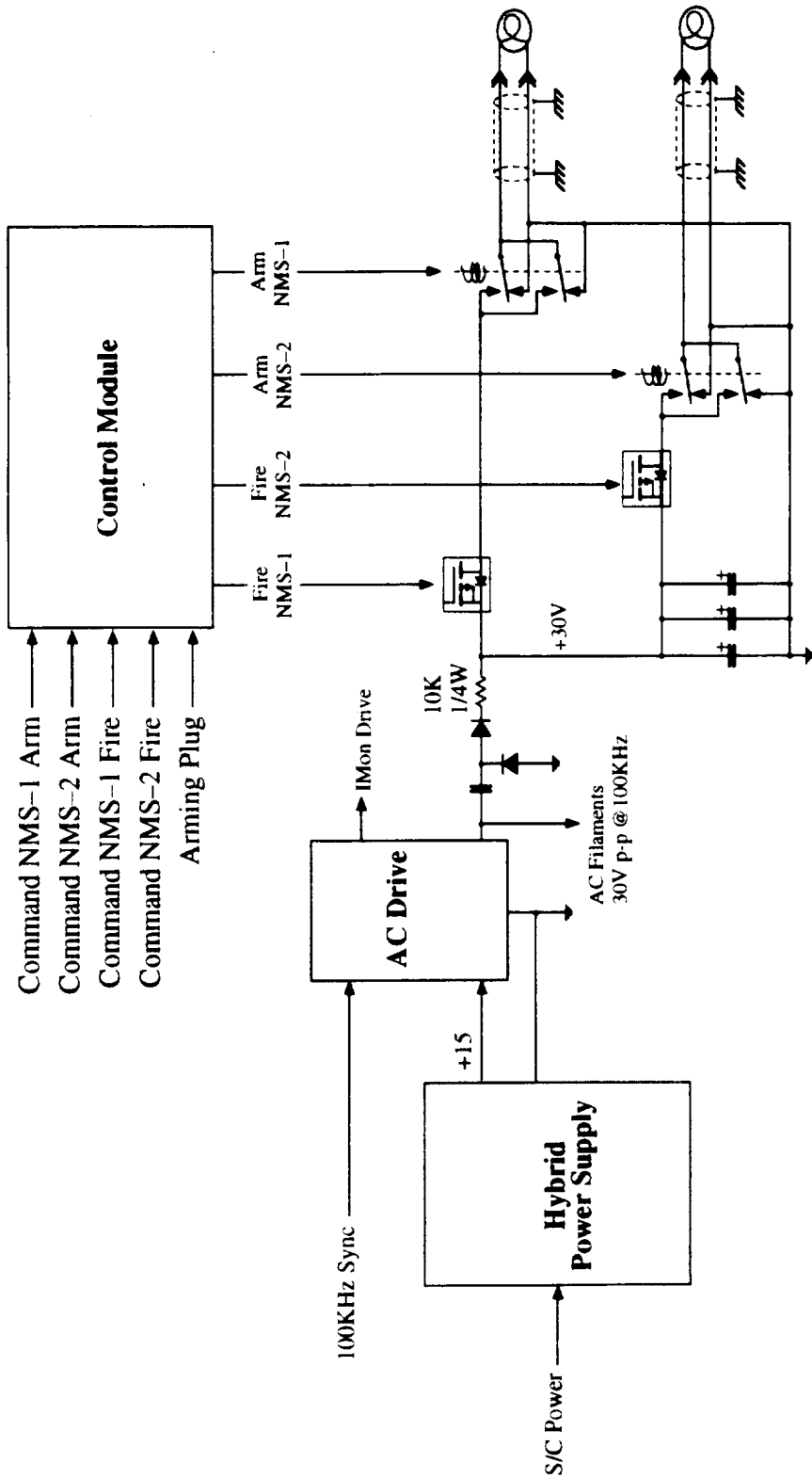
Power Concept
Planet - B NMS

061-NMS 101K63 10110C-Power Concept 061-0110

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1.03





Capacitors:
Sprague
M39006/22-0640
140/60
CLR79 T4

FETs:
IR
IRHF9130

Relays:
Teledyne TO-5
412D-12
M39016/15-129L

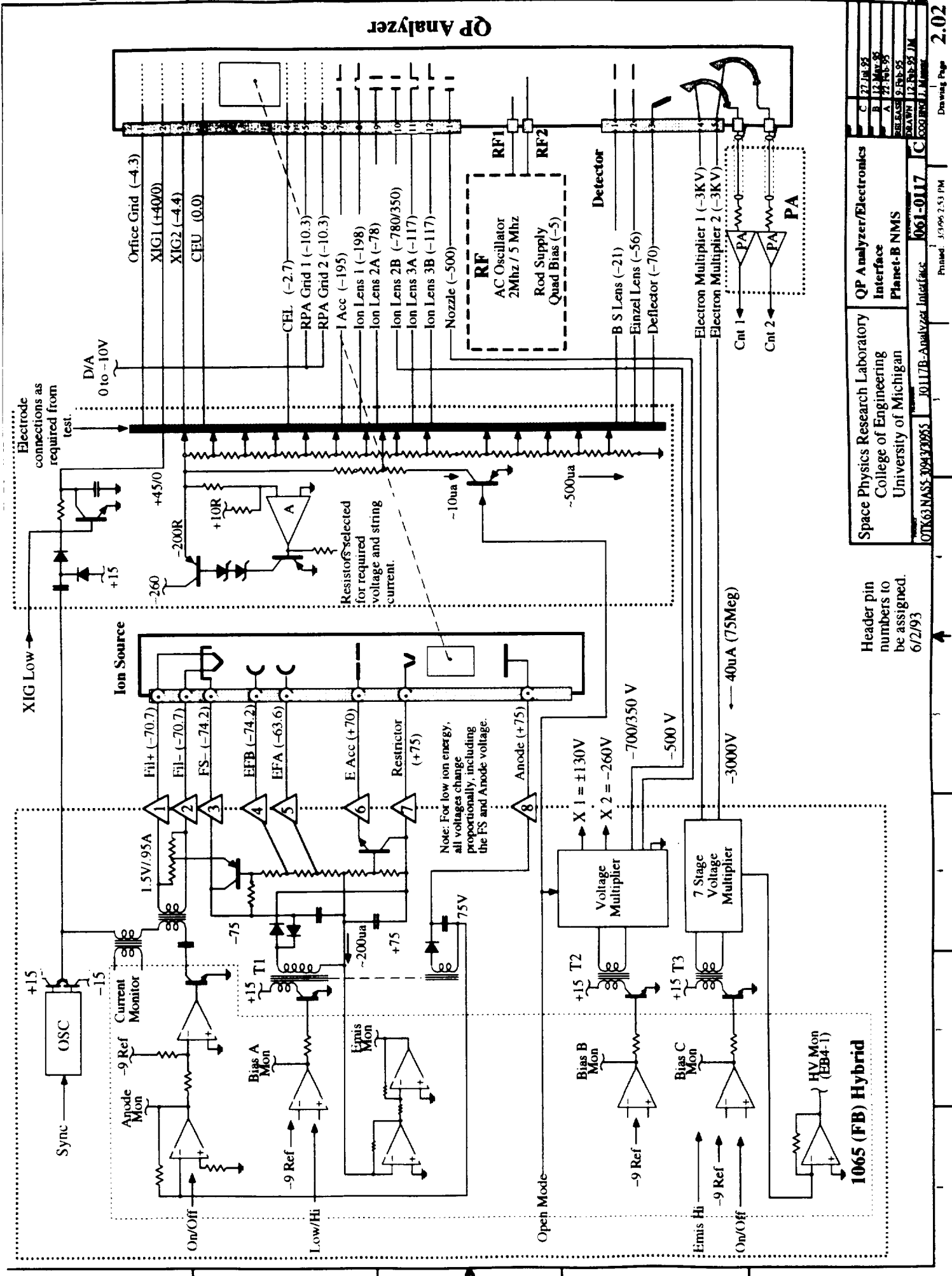
Pyro Device:
IRECO Inc. BA31K7
No Fire = 50 mA
Recommended Fire = 1A
Resistance = 4 to 5 ohms

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University of Michigan

Pyro Concept
Planet-B NMS

061-NMS 101K63 10120B-Pyro Concept 061-0120 B

1.07



Header pin numbers to be assigned. 6/2/93

Safety/Test

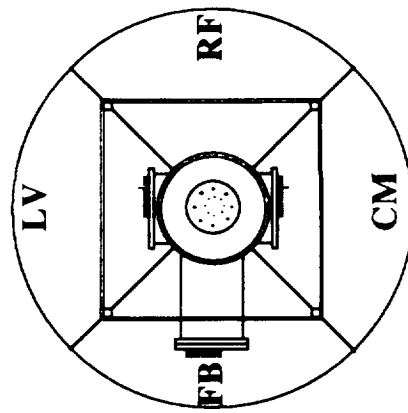
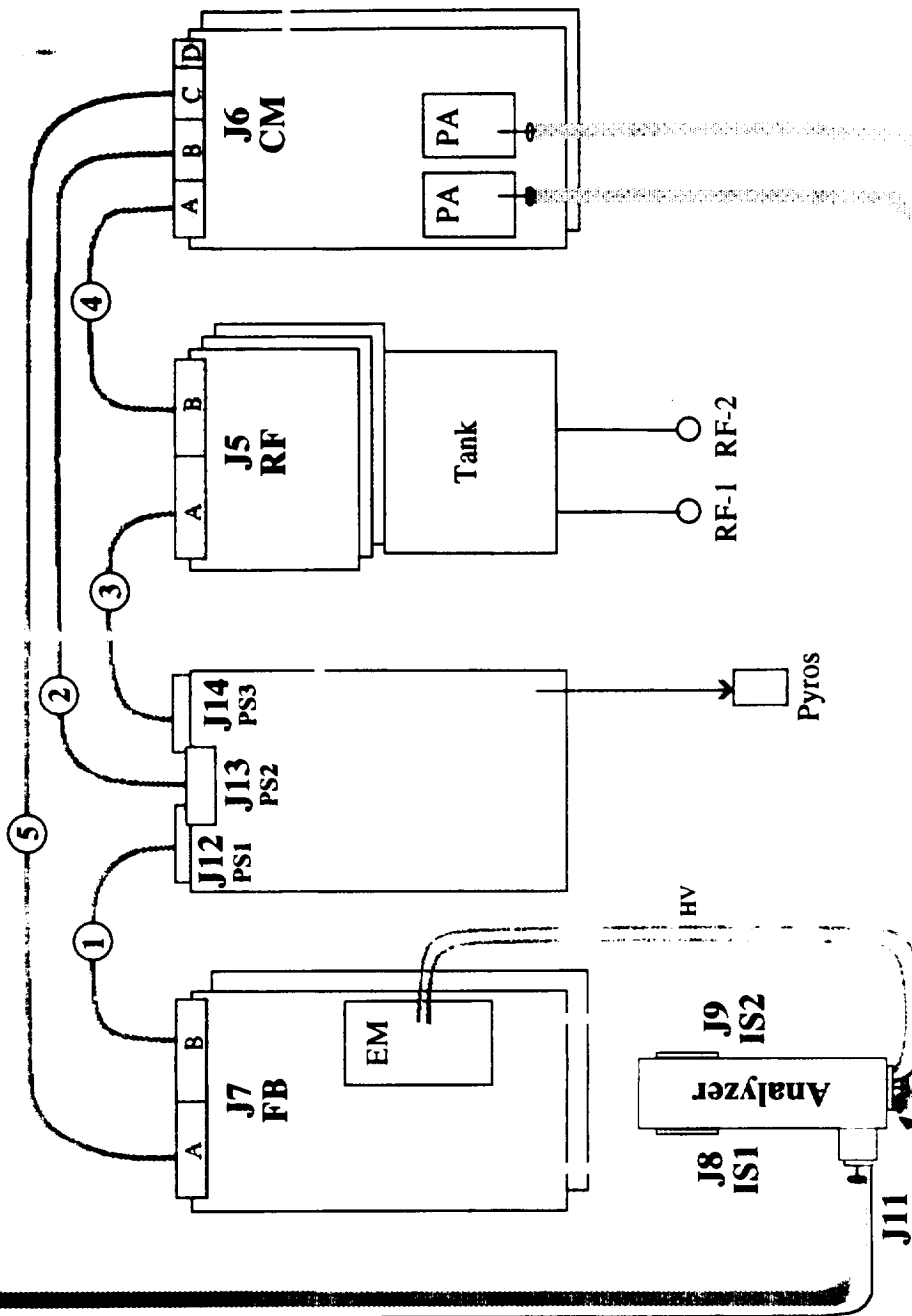
J3
ST
(MDM-15P)

Power/Data

J1
PW
(MDM-15S)

PIM

J2
PM
(MDM-51S)



End View of Analyzer

Note: = Solder connection to PCB.

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University of Michigan

Harness Pictorial
Planet - B NMS

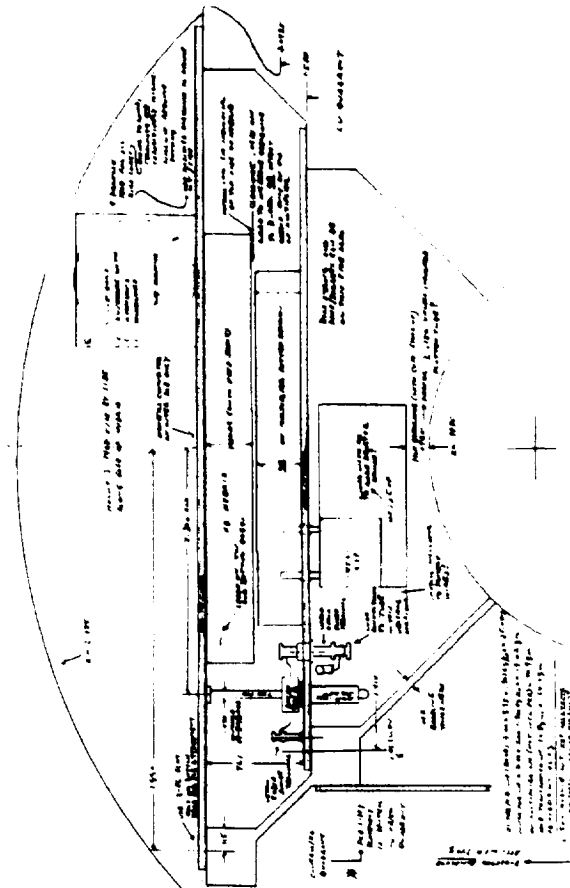
061-NMS 0TK63 J0124A-Harness Pictorial 061-0124 A

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Drawing Page

3.01

DESIGNED	12-08-94
DRAWN	12-08-94
CHECKED	12-08-94
APPROVED	12-08-94
DATE	12-08-94
BY	MDM



Linear
Regulator
Reference

+8.4 Ref

+15

RF +15 [11]

RF 15 Gnd [6]

-RF AGC [1]

RF Temp [1]

RF -15 [1]

RF +130 [4]

RF -130 [4]

RF +5 [2]

RF 5 Gnd [7]

30K YSI

Ref

+5

+5

+5

+5

+5

+5

+5

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+5

10 bit
D/A

Ref

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Ref

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D/A

Ref

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10 bit
D/A

Ref

+5

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DRAWINGS



Planet-B



NMS

Electronic System Drawings

Release 2

4-Mar-96

File

Note: The EGSE is a separate book.

Planet-B NMS Electronic Drawing Index

(061-)

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0.02	0099.2	-	7/27/95	NMS Drawing Index	3.09	0345	A 3/1/96	Harness 5
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Planet-B NMS Electronic Drawing Index

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Release 0, 12-May-95

Initial release of Planet-B, NMS, drawing book. Block and Concept drawings are definitive, to the extent shown, but schematics are incomplete.

Release 0.1, 16-May-95

Revise pyro grounding and returns. Mod. voltage multiplier on FB.

Release 0.2 2-Jun-95

Revise MUX assignments, revise FB pins, add GSE drawing.

Release 0.3 12-Jun-95

Revise FB schematics & Pins, change HV multiplier, redraw CM Block.

Release 0.4 20-Jun-95

Revise Imcon, add harness #s to 3D pictorial, revise wire list and CM interface list, harness pictorial shows passthru for J1 data to CM, FB includes emitter follower for HV drive and added resistor for loop stability.

Release 0.5 27-Jul-95

Move Imcon drive to FB. Revise FB AC drive circuit. Add XIG plus supply and control switch. Revise pyro voltage and circuit. Add socket type and temp pin assignments to 3 interface plugs. Significant revisions to PS and FB block diagrams.

Release 0.6 5-Aug-95

Add CMD and harness 6. Rev Block diagram to show 100 and 300 KHz sync. Add +5 Inhibit to several dwgs. Add ground legend. Add Data Packet assignment drawing.

Release 0.7 23-Aug-95

Many minor revisions. Harness drawings revised.

Release 0.8 6-Sep-95

CM interface pin assignment for Eng model. Safe/Arm plug pins assignment. Minor updates.

Release 0.9 5-Oct-95

Many FB changes to reflect breadboard status. RF drawings added.

Release 1.0 13-Oct-95

Updated most drawings to show a released date. This revision reflects the status of the finished breadboard unit as tested in Japan. Plugs reflect the BB types - not flight numbering.

Release 2.0 Feb-96

Released book for PDR. Proposed flight unit configuration. No effort made to show revision release date on individual drawings.

Communicate all errors, omissions, or modifications to:
Bruce Block 313-764-7439, Bruce_Block@umich.edu
John Maurer 313-764-9463, John_Maurer@umich.edu
FTP to/from: JMAURER@SPRLC.SPRL.UMICH.EDU
E-Mail attachments in BINHEX4, or uuencode for
Microsoft WORD, EXCEL, PROJECTS & AutoCad, & Canvas

Space Physics Research Laboratory
College of Engineering
University of Michigan

Revision Record

Planet - B NMS

061-NMS 0TK63 J0143-Revision Record

061-0143

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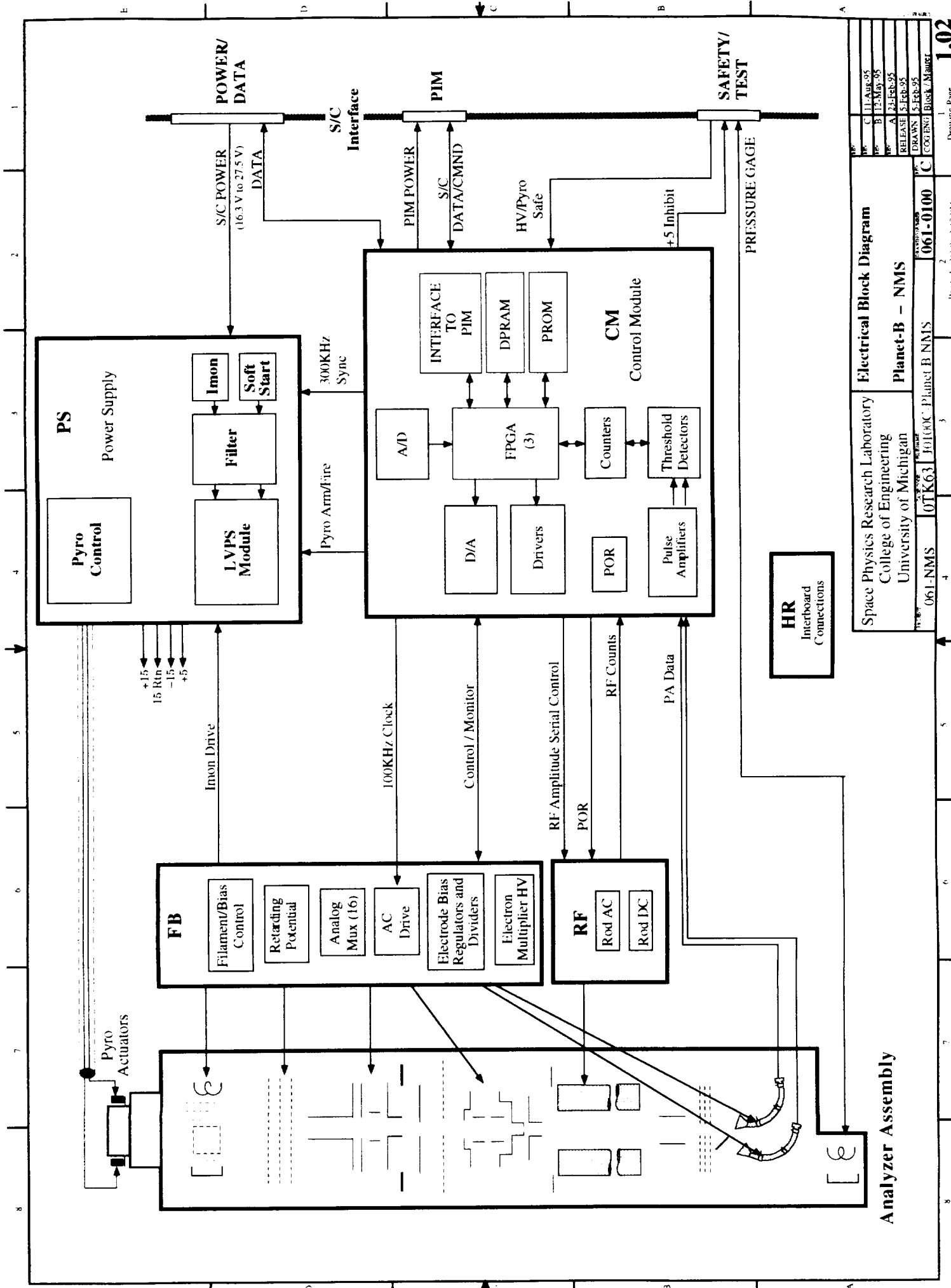
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Section 1

Overview & Concept Documents



Electrical Block Diagram

Planet-B - NMS

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061-NMS 07K63 10100C Planet B NMS

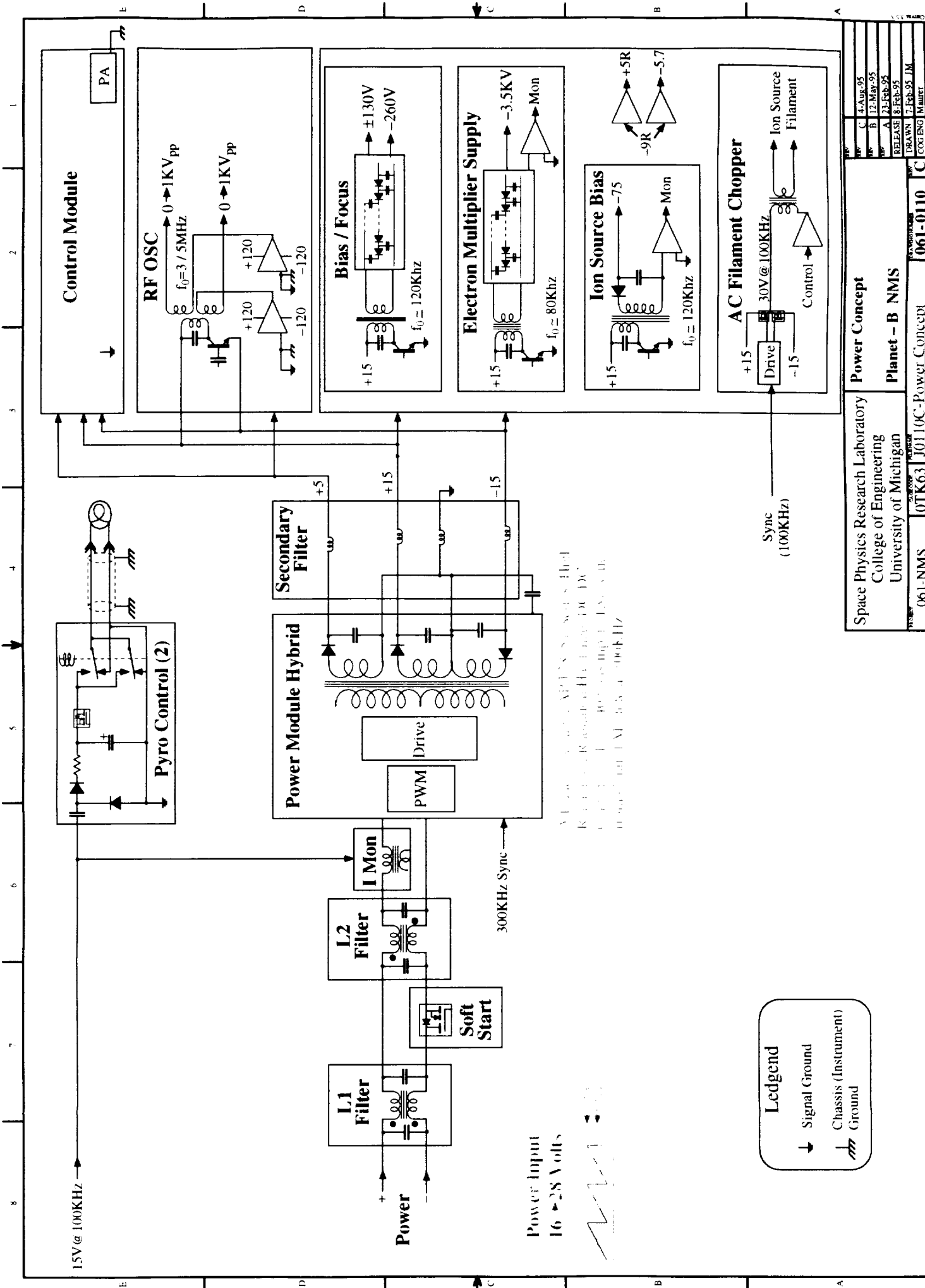
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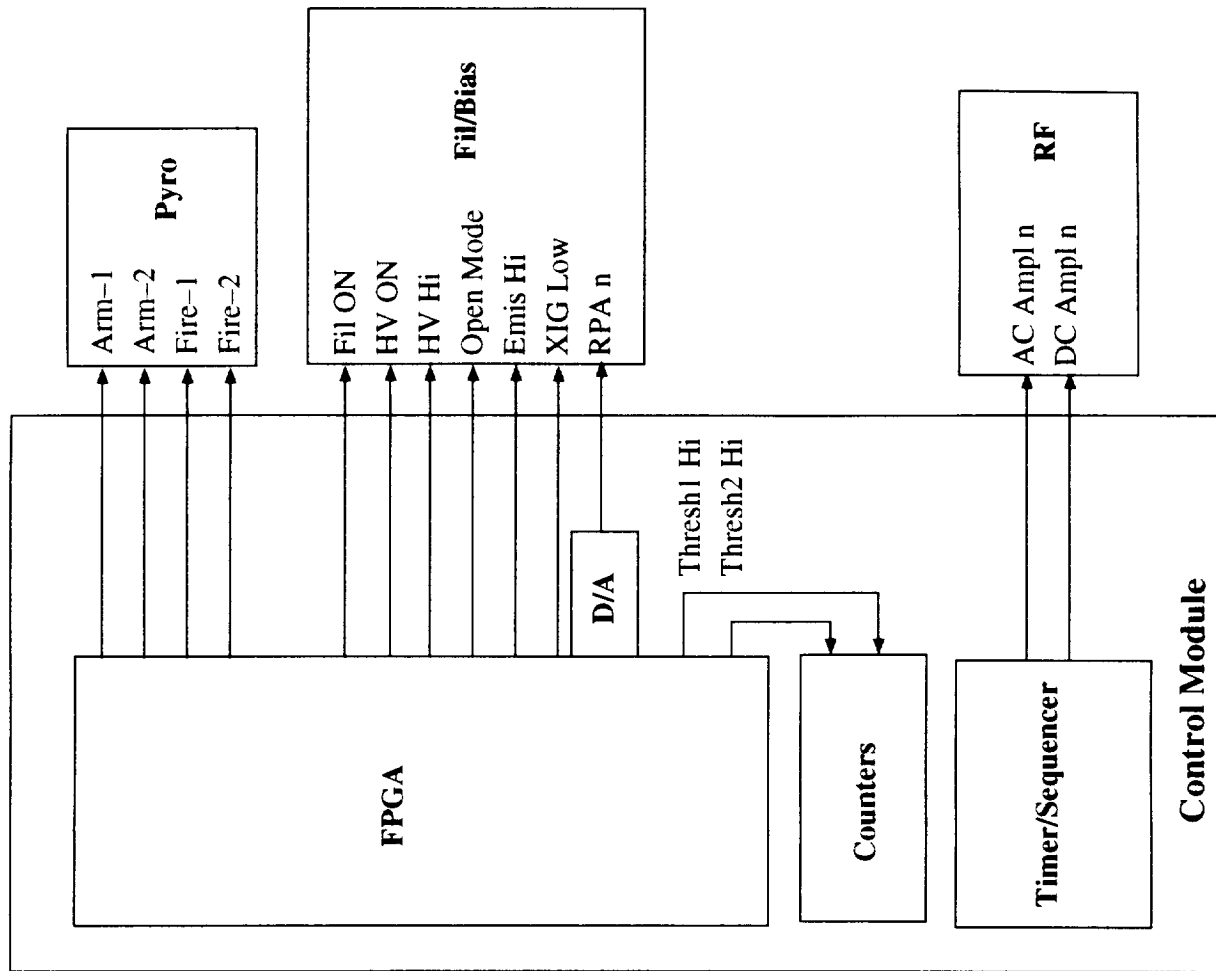
REVISIONS

REV	DATE	DESCRIPTION
1	11-AUG-95	C
2	13-MAY-95	B
3	23-FEB-95	A
4	15-FEB-95	RELEASE
5	15-FEB-95	DRAWN
6	15-FEB-95	CHECKED
7	15-FEB-95	CO-ENG BLOCK / MAINT

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Drawing Page **1.02**





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Planet - B NMS

061-NMS 10TK63 10122A-Comm&Ctrl Concept 061-0122 A

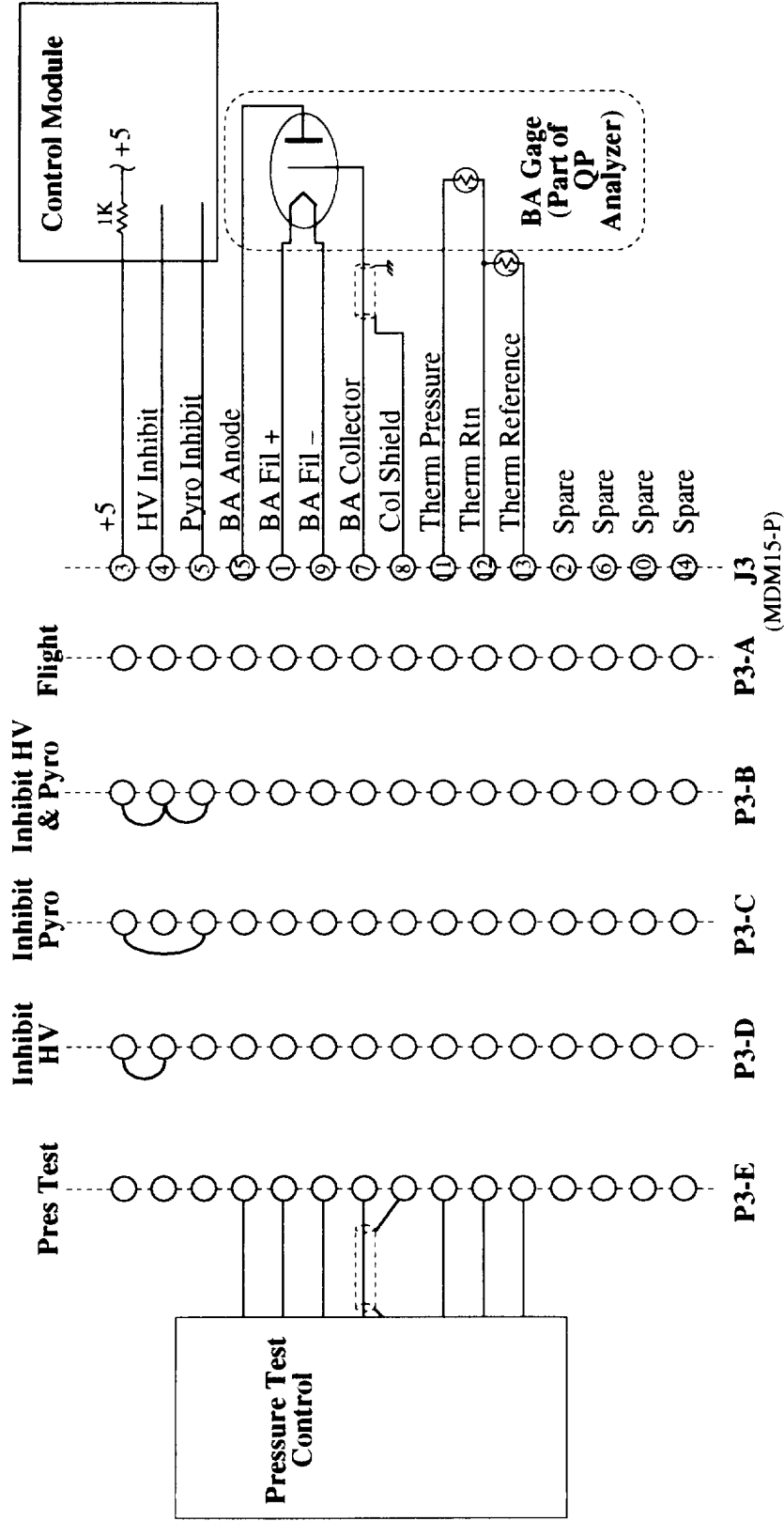
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Drawing Page

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DATE: 20-Jul-95
RELEASE: 12-Mar-95
DRAWN: 18-Apr-95
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Optional Plugs



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University of Michigan

Safe/Arm Concept
Planet - B NMS

061-NMS 0TK63 J0121A-Safe/Arm Concept

061-0121 A

COG ENG Maurer

RELEASE 12-MAY-95

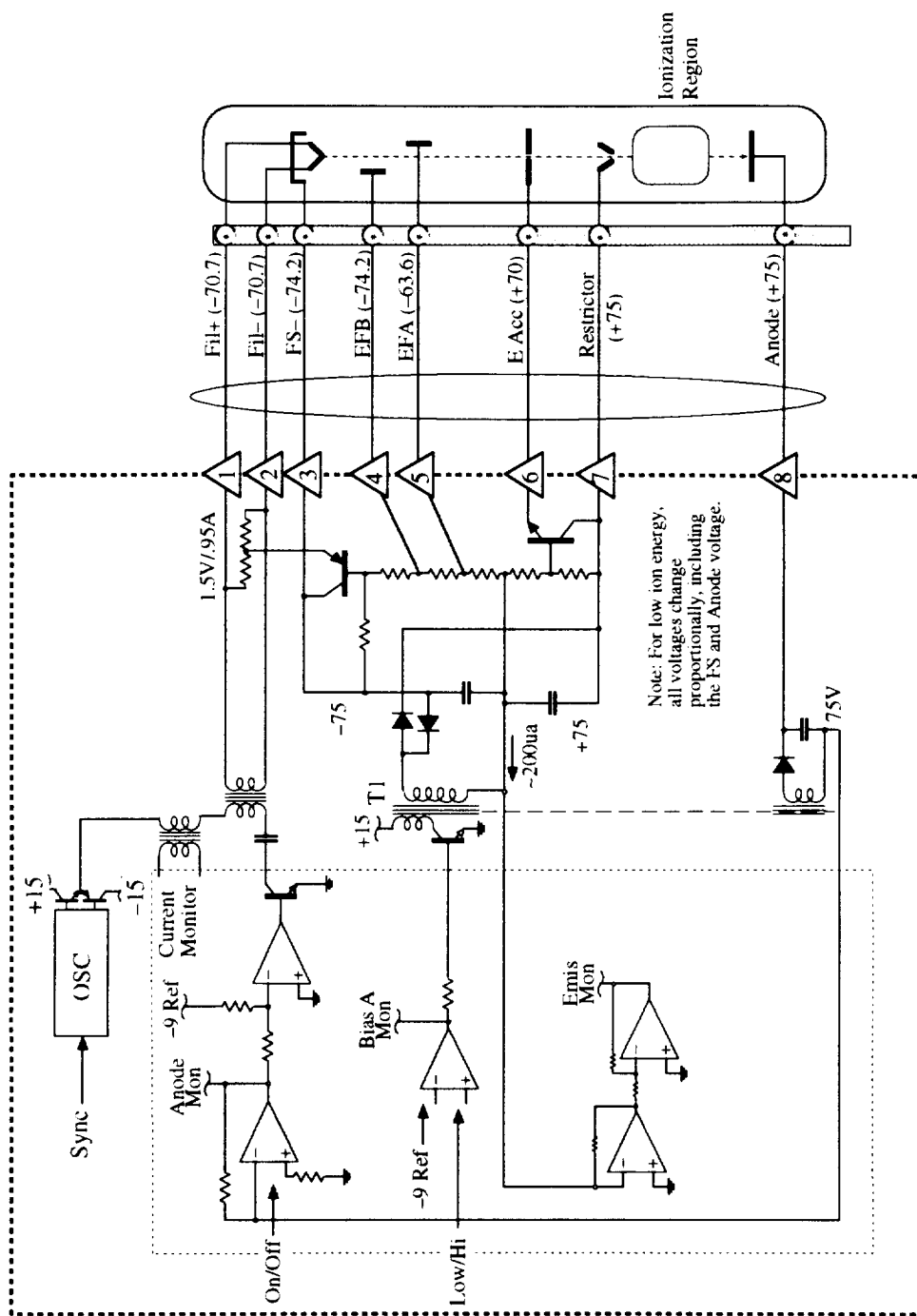
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AMUX Channels in Fil/Bias Module

#	GCMS Function	Source	NMS	Type
1.	Anode 1	Int	Fil Anode	Inv curnt
2.	Anode 2	From RF	-RF AGC*	Inv curnt
3.	EB4 -2	Board	RF Temp	Non-inv curnt
4.	EB4 -1	Board	EM Mon	Non-inv curnt
5.	Fil I 1	Int	Fil I	Direct w/ 5limit
6.	Fil I 2	From LV	PS Temp	Direct w/ 5limit
7.	Fil Emis 2	Board	F/B Temp	Non-inv curnt
8.	Fil Emis 1	Int	Fil Emis	Non-inv curnt
9.	Bias A (-50 dual)	Int	Bias C Mon	Int AC rect
10.	Bias 1 (-70-1)	Int	Bias A Mon	Int AC rect
11.	Bias 2 (-70-2)	Int	Bias B Mon	Int AC rect
12.	Pressure 1	Board	Spare	Curnt w/ gain=100
13.	Bias Mon 3	Board	+5V mon	Direct
14.	Bias Mon 2	From LV	I Mon	Direct
15.	Spare Mon	From DC	Multana1	Direct
16.	Bias Mon 1	From DC	Multana2	Direct

12 is pressure mon with a gain of 100 – either inverting or not.

Space Physics Research Laboratory College of Engineering University of Michigan		Analog Mux Channels	
Planet – B NMS			
061-NMS	0TK63	061-0114	B
DATE: 3/6/1996 12:07 PM		DRAWN: JMB/95	
CHECKED: MAUR		DATE: 23-Aug-95	
		DATE: 26-MAY-95	
		DATE: 1-MAY-95	
		DATE: 1-MAY-95	



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Emission Bias Concept
Planet - B

INMS/NAS53094 0TK63 J0113-Emission Bias Concept 061-0113

Load	Minimum mA			Maximum mA			Watts @72% Input @ 26V	mA Input @ 26V
	15	-15	5	15	-15	5		
RF	15	15	25	230	230	25	9.76	
Control Modul	10	10	100	10	10	100	1.11	
PA	2	25	25	2	25	25	0.74	
Elect. Bias	20	3		20	3		0.48	
Fil Bias	5	1		5	1		0.13	
EM HV	20	1		20	1		0.44	
Filament	0	0	5	75	75	5	3.16	
I mon	3	3		3	3		0.13	
PIM			100			130	0.90	
No PIM Load	75	58	155				3.85	148
Minimum	75	58	255				4.54	175
Max				365	348	285	16.83	647
Min current load	200	259	32					
Resistance Max current load				41	43	18		
Resistance								

FORMAT OF NMS MISSION DATA PACKET			
Data Address	Description	# Bytes	Comments
FA00 .. FA01	Frame Reference Word	2	Binary reference word (fixed 16-bit value, TBD)
FA02 .. F103	Frame Count	2	Count of packet output; increments on Packet Acknowledge
FA04 .. FA06	TI Time	3	Latched on the most recent Index Pulse (IP) after start of first integration period in the current packet
FA07 .. FA08	Phase Count	2	Latched at start of first integration period in the current packet
FA09 .. FA0A	Mem Check Address	2	Address of first memory block being output; increment by 1 each packet
FA0B .. FA1A	Mem Check Data	16	16 bytes of memory data starting at the address above
FA1B .. FA1C	RF Freq 1st	2	RF count sampled in first mass period of the current packet
FA1D .. FA1E	RF Freq Last	2	RF count sampled in last mass period of the current packet
FA1F .. FA20	Sequence Address	2	Memory address of first step in sequence in the current packet
FA21 .. FA22	A Counts	2	Counts from Channel A, mass sample 1
FA23 .. FA24	B Counts	2	Counts from Channel B, mass sample 1
FA25 .. FA26	A Counts	2	Counts from Channel A, mass sample 2
FA27 .. FA28	B Counts	2	Counts from Channel B, mass sample 2
FA29 .. FA2A	A Counts	2	Counts from Channel A, mass sample 3
FA2B .. FA2C	B Counts	2	Counts from Channel B, mass sample 3
FA2D .. FA2E	A Counts	2	Counts from Channel A, mass sample 4
FA2F .. FA30	B Counts	2	Counts from Channel B, mass sample 4
FA31 .. FA32	Analog chan + value	2	4 bit Mux addr + 12 bit A/D sampled after 4th mass samples
FA33 .. FA34	A Counts	2	Counts from Channel A, mass sample 5
FA35 .. FA36	B Counts	2	Counts from Channel B, mass sample 5
FA37 .. FA38	A Counts	2	Counts from Channel A, mass sample 6
FA39 .. FA3A	B Counts	2	Counts from Channel B, mass sample 6
FA3B .. FA3C	A Counts	2	Counts from Channel A, mass sample 7
FA3D .. FA3E	B Counts	2	Counts from Channel B, mass sample 7
FA3F .. FA40	A Counts	2	Counts from Channel A, mass sample 8
FA41 .. FA42	B Counts	2	Counts from Channel B, mass sample 8
FA43 .. FA44	Analog chan + value	2	4 bit Mux addr + 12 bit A/D sampled after 8th mass samples
FA45 .. FA6A	Data of 2nd 8-Samples	38	Above block repeats nine more times
FA6B .. FA90	Data of 3rd 8-Samples	38	
FA91 .. FAB6	Data of 4th 8-Samples	38	
FAB7 .. FADC	Data of 5th 8-Samples	38	
FADD .. FB02	Data of 6th 8-Samples	38	
FB03 .. FB28	Data of 7th 8-Samples	38	
FB29 .. FB4E	Data of 8th 8-Samples	38	
FB4F .. FB74	Data of 9th 8-Samples	38	
FB75 .. FB9A	Data of 10th 8-Samples	38	
FB9B .. FBAF	Spare	21	
TOTAL		432	Total 8-bit bytes in NMS packet. 1 NMS packet/sec at maximum data rate.

Value/Generic Pt		Part Number	Mfr.	Grade/		Re-	PARTS BY DECK					Flight Parts							Stock					
				NSPAR	Sern.		PS	FB	RF	MB	HR	Tot	Qty	DPA		Price	Ord	Data		Total	SPR	Qty.	Comment	
														Req	Qty									Chg
CAPACITORS																								
1000p/200 CKR05	AVX	M39014/01-1357	2	N	0	0	1				1	2	0	2	0.35	10	0	3.50	77	GCMS				
220p/200 CKR05	Kem	M39014/01-1345	2	N	0	0	4				4	8	0	8	0.30	15	0	4.50	242	GCMS				
47P/200 CKR05	Kem	M39014/01-1333	2	N	0	0	1				1	2	0	2	0.84	10	0	8.40	23	GCMS				
1/50 CKR05	Kem	M39014/01-1593	2	N	0	0	12				12	24	0	24	0.30	25	0	7.50	205	GCMS				
4700P/100 CKR05	Kem	M39014/01-1569	2	N	0	0	2				2	4	0	4	0.54	20	0	10.80	20	GCMS				
01/100 CKR05	Kem	M39014/01-1575	2	N	9	0	17				26	52	0	52	0.30	70	0	21.00	80	GCMS				
3900p/100 CKR05	Kem	M39014/01-1568	2	N	0	0	1				1	2	0	2	0.31	10	0	3.10	145	GCMS				
3300/50 CKR05					0	1	0																	
3300/100 CKR05					0	2	0																	
01/200 CKR06	AVX	M39014/02-1338	2	N	0	2	0				2	4	0	4	0.35	25	0	8.75	57	GCMS				
33/50 CKR06	AVX	M39014/02-1358	2	N	2	0	0				2	4	0	4	0.60	10	0	6.00	90	GCMS				
1/100 CKR06	Kem	M39014/02-1350	2	N	12	13	0				25	50	0	50	0.33	70	0	23.10	291	GCMS				
330P/100 CCR05	Kem	CCR05CG331FS	1	N	0	0	0				0	0	0	0	1.37	10	0	13.70	50	GCMS				
330/200 CCR05					0	1	0																	
330/500 HV21					0	1	0																	
33/15 CLR79	SPG	M39006/22-0505	1	N	0	2	2				4	8	0	8	8.92	15	0	133.80	11	GCMS				
15/30 CLR79	SPG	M39006/22-0545	1	N	0	1	0				1	2	0	2	7.96	10	0	79.60	22	GCMS				
10/50 CLR79	SPG	M39006/22-0565	1	N	0	2	0				2	4	0	4	7.96	10	0	79.60	36	GCMS				
8/260 CLR79	SPG	M39006/22-0585	1	N	0	1	0				1	2	0	2	8.92	15	0	133.80	34	GCMS				
47/35 CLR79					0	0	0																	
56/75 CLR79	SPG	M39006/22-0617	1	N	1	0	0				1	2	0	2	12.65	10	0	126.50	35	GCMS				
140/60 CLR79	SPG	M39006/22-0640	1	N	3	0	0				3	6	0	6	21.25	10	0	212.50	0					
17075 CLR79T2					0	1	0																	
22/35 CSR21C					2	0	0																	
22/35 CSR21D	SPG	M39003/09-0353	1	N	0	2	0				2	4	0	4	3.77	10	0	37.70	17	TOMS				
220/10 CSR21D	SPG	M39003/09-0321	1	N	1	0	0				1	2	0	2	7.47	10	0	74.70	10	TOMS				
015/2KV HS20 NP	KD	20HS20N825KMCS	#014	N	0	0	0				0	0	0	0	88.71	20	0	1774.20	0					
047/1KV HS20 NP	KD	20HS20N825KMCS	#014	N	0	0	0				0	0	0	0	88.71	10	0	887.10	0					
082/500 HS20 NP	KD	20HS20N825KMCS	#014	N	0	0	0				0	0	0	0	88.71	15	0	1330.65	0	GCMS				
22/200 HV21	KD	02HV21B224KMS	#015	N	0	1	0				1	2	0	2	91.90	10	0	919.00	26					
22/200 HV22					0	5	0																	
39/400V HV23	KD	04HV23B394KMS	#015	N	2	0	0				2	4	0	4	88.34	30	0	2650.20	14	GCMS				
0.015/2KV HV23					0	7	0																	
0.15/800 HV23					0	1	0																	
0.047/1KV HV21					0	4	0																	

Value/Generic Pt		Part Number	Mfr.	Grade/ NSPAR	Re- Scrn.	SCD	PARTS BY DECK				Flight Parts							Stock																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																															
							PS	FB	RF	MB	HR	Tot	Qty Req'	DPA Qty	Tot Qty	Price Each	Ord Qty		Data Chg	Total Cost	SPR Qty.	Comment																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																											
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Value/Generic Pt		Part Number	Mfr.	Grade/ NSPAR	Re- Sern.	PARTS BY DECK					Flight Parts										
						PS	FB	RF	MB	HR	Tot	Qty Req	DPA Tot		Price Each	Ord Data		Total Cost	SPR Qty.	Stock Comment	
													Qty	Qty		Qty	Chg				
IN4148-1		JANSIN4148-1	MS	1	N	N	1	10	1			12	18	5	23	32.81	25	0	820.25	13	GCMS
IN751A-1		JANTXVIN751A-1	MS	2	Y	N	0	1	0			1	2	5	7	47.88	10	0	478.80	32	GCMS
IN4995		JANTXVIN4995	MS	2	N	N	0	1	0			1	2	5	7	32.14	10	0	321.40	0	
IN5617		JANSIN5617	MS	1	N	N	0	2	0			2	4	5	9	25.38	12	0	304.56	15	GCMS
IN5711		JANTXVIN5711	HP	1	N	N	2	0	3			5	8	5	13	5.68	18	0	102.24	6	GCMS
IN5806		JANSIN5806	MS	1	N	N	5	0	0			5	8	5	13	41.42	15	0	621.30	45	GCMS
IN5811		JANSIN5811	MS	1	N	N	2	0	0			2	4	5	9	43.88	12	0	526.56	32	GCMS
IN4569A		JANSIN4569A	MS	1	N	N	0	0	1			1	2	5	7	26.11	12	0	313.32	0	
IN6320		JANSIN6320	MS	1	N	N	0	0	0			0	0	5	5	30.03	12	0	360.36	21	GCMS
IN6325		JANSIN6325	MS	1	N	N	1	0	0			1	2	5	7	30.03	12	0	360.36	9	GCMS
HRF30		G311P670-HRF30	SEM	#009	Y	S-311	0	28	0			28	42	5	47	27.44	40	0	1097.60	21	GCMS
IN754A		JANSIN754	MS	1	N	N	1	0	0			1	2	5	7	30.03	10	0	300.30	0	
HYBRID MICROCIRCUITS																					
ART2815/883		ART2815/883	AAD	#062	N	N	0	0	0			0	0	0	0	18000.00	3		54000.00	0	
A-E10156		A-E10156	Tel	NA	N	N	0	0	1			1	2	0	2	-	-		0.00	0	
A-E10198		A-E10198	Tel	NA	N	N	0	1	0			1	2	0	2	-	-		0.00	0	
A-E10935		A-E10935					0	0	1												
A-E10936		A-E10936					0	0	1												
A-E10937		A-E10937					0	0	1												
INDUCTORS																					
33 uH		M39010/03-A330JP	NYT	1	N	N	0	0	0			0	0	0	0	12.83	10	0	128.30	11	GCMS
33uH		M39010/02					0	0	1												
TRANSFORMERS																					
M21038		M21038/05-27	TRL	1	N	N	1	0	0			1	2	0	2	60.00	5	0	300.00	5	GCMS
RELAYS																					
412D-12		M39016/15-079P	TEL	#064	N	N	2	0	0			2	4	0	4	42.61	10	0	426.10	6	TOMS
TRANSISTORS																					
2N3632A							0	0	1												

Value/Generic Pt		Part Number	Mfr.	Grade/ NSPAR	Re- Sern.	PARTS BY DECK				Flight Parts										
						PS	FB	RF	MB	HR	Tot	Qty Req	DPA Qty	Tot Qty	Price Each	Ord Qty	Chg	Total Cost	SPR Qty.	Stock
2N2222A	JANS2N2222A	Mot	1	N	N	0	2	7		9	14	5	19	42.20	25	0	1055.00	30	GCMS	
2N2907A	JANS2N2907A	Mot	1	N	N	0	5	1		6	9	5	14	43.00	15	0	645.00	11	GCMS	
2N3700	JANS2N3700	Mot	1	N	N	4	7	0		11	17	5	22	43.90	25	0	1097.50	35	GCMS	
2N3866	5961012352202	Mot	2	Y	N	0	0	1		1	2	5	7	61.62	12	0	739.44	19		
SFT1192/5	SFT1192	SSD	???	Y	N	0	4	0		4	6	5	11	10.37	15	0	155.55	0		
IRHF9130	IRHF9130	IR	#061	N	N	2	0	0		2	3	5	8	279.95	10	0	2799.50	0		
IRFM254	91-0072	IR	#001	N	N	1	0	0		1	2	5	7	418.47	10	0	4184.70	10	TOMS	
AE11076	TBD	TBD	2	Y	N	0	0	0		0	0	5	5	53.78	10	0	537.80	0		
2N2369	JANTXV2N2369	Mot	2	Y	N	1	0	0		1	2	5	7	15.36	10	0	153.60	0		
RESISTORS																				
100 RCR05	RCR05G101JS	AB	1	N	N	1	0	8		9	18	0	18	0.60	50	0	30.00	0	TBD	
1K RCR05	RCR05G102JS	AB	1	N	N	1	1	2		4	8	0	8	0.60	50	0	30.00	0	TBD	
10K RCR05	RCR05G103JS	AB	1	N	N	4	3	4		11	22	0	22	0.60	50	0	30.00	0	TBD	
100K RCR05	RCR05G104JS	AB	1	N	N	0	0	0		0	0	0	0	0.69	50	0	34.50	0	TBD	
1M RCR05	RCR05G105JS	AB	1	N	N	0	1	1		2	4	0	4	0.69	50	0	34.50	0	TBD	
1.2K RCR05	RCR05G122JS	AB	1	N	N	0	0	5		5	10	0	10	0.69	50	0	34.50	0	TBD	
1.5K RCR05	RCR05G152JS	AB	1	N	N	0	0	2		2	4	0	4	0.25	50	0	12.50	0	TBD	
150 RCR05	RCR05G151JS	AB	1	N	N	0	0	0		0	0	0	0	0.28	50	0	14.00	0	TBD	
15K RCR05	RCR05G153JS	AB	1	N	N	2	1	0		3	6	0	6	0.28	50	0	14.00	0	TBD	
20K RCR05	RCR05G203JS	AB	1	N	N	0	0	0		0	0	0	0	0.60	50	0	30.00	0	TBD	
200K RCR05	RCR05G204JS	AB	1	N	N	0	0	0		0	0	0	0	0.69	50	0	34.50	0	TBD	
2M RCR05	RCR05G205JS	AB	1	N	N	0	0	0		0	0	0	0	0.69	50	0	34.50	0	TBD	
220 RCR05	RCR05G221JS	AB	1	N	N	0	0	1		1	2	0	2	0.69	50	0	34.50	0	TBD	
2.2K RCR05	RCR05G222JS	AB	1	N	N	1	0	0		1	2	0	2	0.69	50	0	34.50	0	TBD	
2.4K RCR05						0	0	1		1	2	0	2							
22K RCR05	RCR05G223JS	AB	1	N	N	0	0	1		1	2	0	2	0.69	50	0	34.50	0	TBD	
2.7K RCR05	RCR05G272JS	AB	1	N	N	0	0	0		0	0	0	0	0.25	50	0	12.50	0	TBD	
270K RCR05	RCR05G274JS	AB	1	N	N	0	0	0		0	0	0	0	0.69	50	0	34.50	0	TBD	
33K RCR05	RCR05G273JS	AB	1	N	N	1	0	0		1	2	0	2	0.28	50	0	14.00	0	TBD	
330 RCR05						0	0	1		1	2	0	2							
390K RCR05						0	0	1		1	2	0	2							
3.3K RCR05	RCR05G332JS	AB	1	N	N	0	0	1		1	2	0	2	0.25	50	0	12.50	0	TBD	
3.6K RCR05						0	0	1		1	20	0	20	0.60	50	0	30.00	0	TBD	
4.7K RCR05	RCR05G472JS	AB	1	N	N	5	2	3		10	20	0	20							
47 RCR05						0	1	0		0										

PLANET B FLIGHT PARTS COST/PROCUREMENT STATUS SCHEDULE

Value/Generic Pt	Part Number	Mfr.	Grade/ NSPAR	Re- Scrn.	PARTS BY DECK					Flight Parts					Stock Comment					
					PS	FB	RF	MB	HR	Tot	Qty Req'	DPA Qty	Tot Qty	Price Each		Ord Qty	Data Chg	Total Cost	SPR Qty.	
																				SCD
470 RCR05	RCR05G471JS	AB	1	N	N	5	0	0				10	0	10	0.60	50	0	30.00	0	TBD
4.3K RCR05	RCR05G432JS	AB	1	N	N	0	0	0				0	0	0	0.60	50	0	30.00	0	TBD
43K RCR05	RCR05G433JS	AB	1	N	N	0	0	1				2	0	2	0.60	50	0	30.00	0	TBD
47K RCR05	RCR05G473JS	AB	1	N	N	5	4	0				18	0	18	0.60	50	0	30.00	0	TBD
330 RCR05																				
510 RCR05	RCR05G511JS	AB	1	N	N	0	1	0				2	0	2	0.60	50	0	30.00	0	TBD
5.5KRCR05																				
56K RCR05	RCR05G563JS	AB	1	N	N	0	0	0				0	0	0	0.28	50	0	14.00	0	TBD
620RCR05																				
6.2K RCR05	RCR05G622JS	AB	1	N	N	0	0	1				2	0	2	0.69	50	0	34.50	0	TBD
68 RCR05	RCR05G680JS	AB	1	N	N	0	1	1				4	0	4	0.69	50	0	34.50	0	TBD
680 RCR05	RCR05G681JS	AB	1	N	N	0	2	2				8	0	8	0.48	50	0	24.00	0	TBD
7.5K RCR05	RCR05G752JS	AB	1	N	N	0	0	1				2	0	2	0.69	50	0	34.50	0	TBD
82 RCR05	RCR05G82RJS	AB	1	N	N	0	0	1				2	0	2	0.48	50	0	24.00	0	TBD
8.2K RCR05	RCR05G822JS	AB	1	N	N	0	0	1				2	0	2	0.48	50	0	24.00	0	TBD
9.1K RCR05	RCR05G912JS	AB	1	N	N	0	0	2				4	0	4	0.48	50	0	24.00	0	TBD
1K RCR07																				
10K RCR07	RCR07G103JR	AB	2	N	N	0	2	0				8	0	8	0.17	50	0	8.50	0	TBD
2M RCR07	RCR07G205JR	AB	2	N	N	0	1	0				2	0	2	0.17	50	0	8.50	0	TBD
2.7 RCR07	RCR07G2R7JR	AB	2	N	N	1	0	0				2	0	2	0.17	50	0	8.50	0	TBD
47 RCR07																				
100K RCR07	RCR07G104JR	AB	2	N	N	0	2	0				4	0	4	0.17	50	0	8.50	0	TBD
10 RCR07	RCR07G100JR	AB	2	N	N	0	0	0				0	0	0	0.17	50	0	8.50	0	TBD
51 RCR07	RCR07G510JR	AB	2	N	N	0	0	0				0	0	0	0.37	50	0	18.50	0	TBD
110 RCR07	RCR07G111JR	AB	2	N	N	2	0	0				4	0	4	0.37	50	0	18.50	0	TBD
499K RCR07																				
110RNC50																				
1.00K RNC50	RNC50J1001FS	Dal	1	N	N	0	1	0				0	0	0	0.24	50	0	12.00	0	TBD
10.0K RNC50	RNC50J1002FS	Dal	1	N	N	0	0	0				0	0	0	0.59	50	0	29.50	0	TBD
100K RNC50	RNC50J1003FS	Dal	1	N	N	1	2	2				10	0	10	0.59	50	0	29.50	0	TBD
120 RNC50																				
110K RNC50																				
12.1K RNC50	RNC50J1212FS	Dal	1	N	N	0	3	0				0	0	0	0.59	50	0	29.50	0	TBD
13.3K RNC50	RNC50J1332FS	Dal	1	N	N	0	0	0				4	0	4	0.59	50	0	29.50	0	TBD
47.5K RNC50	RNC50J4752FS	Dal	1	N	N	0	0	1				2	0	2	0.59	50	0	29.50	0	TBD
15.0K RNC50	RNC50J1502FS	Dal	1	N	N	0	0	0				0	0	0	0.59	50	0	29.50	0	TBD

PLANET B FLIGHT PARTS COST/PROCUREMENT STATUS SCHEDULE

2/23/96

Value/Generic Pt		Part Number	Mfr.	Grade/ NSPAR	Re- Scrn.	PARTS BY DECK				Flight Parts										
						PS	FB	RF	MB	HR	Tot	Qty Req'	DPA		Price Each	Ord Qty	Data Chg	Total Cost	SPR Qty.	Stock Comment
													Qty	Qty						
150K RNC50		RNC50J1503FS	Dal	1	N	N	0	0	0			0	0	0	0.59	50	0	29.50	0	TBD
162K RNC50							0	1	0											
165K RNC50							1	0	0											
17.8K RNC50		RNC50J1782FS	Dal	1	N	N	0	0	0			0	0	0	0.59	50	0	29.50	0	TBD
1.78K RNC50							1	0	0											
1.87K RNC50		RNC50J1871FS	Dal	1	N	N	0	0	0			0	0	0	0.59	50	0	29.50	0	TBD
18.7K RNC50		RNC50J1872FS	Dal	1	N	N	0	3	0			3	6	0	0.59	50	0	29.50	0	TBD
20.0K RNC50		RNC50J2002FS	Dal	1	N	N	0	0	0			0	0	0	0.59	50	0	29.50	0	TBD
200K RNC50							0	1	0											
21.0K RNC50		RNC50J2102FS	Dal	1	N	N	0	0	0			0	0	0	0.59	50	0	29.50	0	TBD
22K RNC50							0	3												
24.9K RNC50		RNC50J2492FS	Dal	1	N	N	0	0	3			3	6	0	0.59	50	0	29.50	0	TBD
16.5K RNC50		RNC50J1652FS	Dal	1	N	N	0	0	2			2	4	0	0.24	50	0	12.00	0	TBD
2.67K RNC50		RNC50J2671FS	Dal	1	N	N	0	0	0			0	0	0	0.59	50	0	29.50	0	TBD
26.7K RNC50							0	0	1											
27K RNC50							0	1	0											
332K RNC50		RNC50J3323FS	Dal	1	N	N	0	0	2			2	4	0	1.76	50	0	88.00	0	TBD
402 RNC50		RNC50J4020FS	Dal	1	N	N	0	0	0			0	0	0	0.59	50	0	29.50	0	TBD
453 RNC50		RNC50J4530FS	Dal	1	N	N	0	0	0			0	0	0	0.59	50	0	29.50	0	TBD
499 RNC50		RNC50J4990FS	Dal	1	N	N	0	0	0			0	0	0	0.59	50	0	29.50	0	TBD
4.02K RNC50							0	0	1											
4.99K RNC50		RNC50J4991FS	Dal	1	N	N	0	0	3			3	6	0	0.59	50	0	29.50	0	TBD
49.9K RNC50		RNC50J4992FS	Dal	1	N	N	0	2	9			11	22	0	0.59	50	0	29.50	0	TBD
499K RNC50		RNC50J4993FS	Dal	1	N	N	0	2	0			2	4	0	1.10	50	0	55.00	0	TBD
56.2K RNC50		RNC50J5622FS	Dal	1	N	N	0	0	0			0	0	0	0.59	50	0	29.50	0	TBD
60.4K RNC50		RNC50J6042FS	Dal	1	N	N	0	0	0			0	0	0	0.59	50	0	29.50	0	TBD
68.1K RNC50		RNC50J6812FS	Dal	1	N	N	0	0	0			0	0	0	0.59	50	0	29.50	0	TBD
825 RNC50		RNC50J8250FS	Dal	1	N	N	0	0	0			0	0	0	0.28	50	0	14.00	0	TBD
8.45K RNC50							0	0	1											
90.9K RNC50		RNC50J9092FS	Dal	1	N	N	0	2	0			2	4	0	0.59	50	0	29.50	0	TBD
93.1K RNC50							0	1	0											
976K RNC50							0	1	0											
226K RNC55							0	1	0											
49.9K RNC55							0	0	2											
30K Thermistor	S-311-P-18-09T101	YSI	1	N	S-311	S-311	1	1	1			3	6	0	53.35	15	0	800.25	7	GCMS
5M MG660	G311P683 660-	Cad	1	N	S-311	S-311	0	1	?			###	###	0	16.50	10	0	165.00	15	GCMS
10M MG660	G311P683 660-	Cad	1	N	S-311	S-311	0	1	?			###	###	0	17.58	10	0	175.80	0	TBD

File: J0380-Flight Parts

Combined Parts Summary
Planet-B NMS

061-0380 Rev. A 15-Feb-96

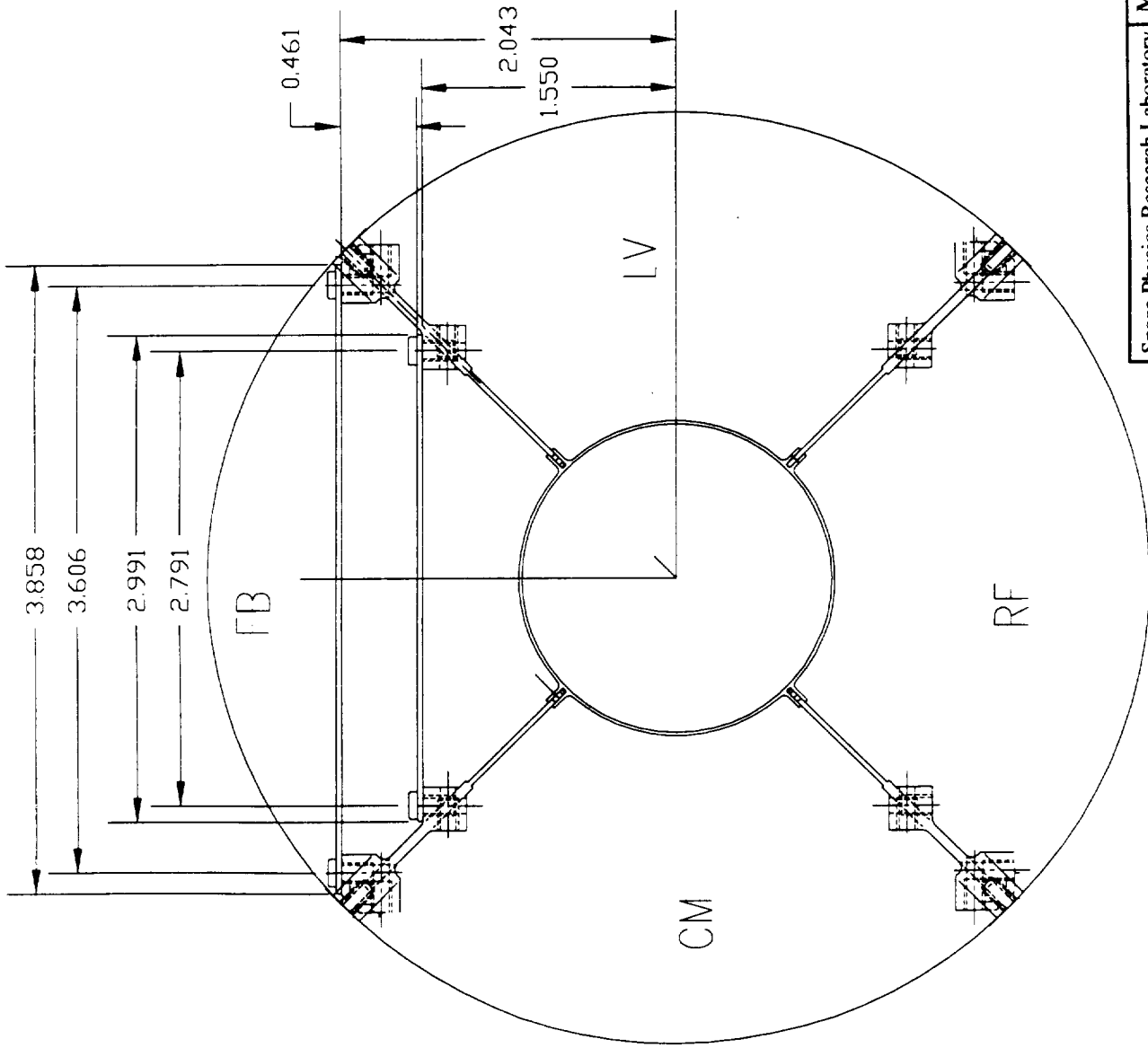
Printed: 2/23/96

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1.25

2/23/96

Value/Generic Pt	Part Number	Mfr.	Grade/ NSPAR	Re- Sern.	PARTS BY DECK					Flight Parts									
					PS	FB	RF	MB	HR	Tot	Qty Req'	DPA Qty	Tot Qty	Price Each	Ord Qty	Data Chg	Total Cost	SPR	Stock
50M MOX400																			
MICROCIRCUITS																			
LT1013	RH1013M	LT	#019	N	Y	0	2	8		10	15	0	15	378.00	15	0	5670.00	18	GCMS
UC1709J/883BC	UC1709J/883BC	UNI	#060	N	N	0	1	0		1	2	0	2	23.90	10	0	239.00	0	
DAC-312	DAC312/883	AD	#059	N	Y	0	0	2		2	4	0	4	51.90	6	0	311.40	0	
54HCS164	54HCS164	Har	1	N	N	0	0	4		4	8	0	8	156.00	6	0	936.00	0	
54HCS273	54HCS273	NS	1	N	N	0	0	8		8	16	0	16	162.00	10	0	1620.00	0	
MISC.																			
Pin	09-8157-1-03	Con	NA	N	N	0	0	0		0	0	0	0	0.39	200	0	78.00	124	GCMS
Socket	1-332070-7	AMP	NA	N	N	0	0	0		0	0	0	0	0.11	2000	0	224.32	1245	GCMS
0-80 Lug Terminal	R26201	Hol	NA	N	N	0	0	0		0	0	0	0	0.05	1000	0	47.50	23	GCMS
2030 Terminal	2030B-45	USE	NA	N	N	0	0	0		0	0	0	0	0.15	2500	0	379.13	450	GCMS
2031 Terminal	2031B-45	USE	NA	N	N	0	55	21		76	152	0	152	0.12	1500	0	183.06	120	GCMS
2061 Terminal	2061B-45	USE	NA	N	N	8	15	0		23	46	0	46	0.21	1000	0	205.20	452	GCMS
1624 Terminal	1624-4-10	USE	NA	N	N	0	0	12		12	24	0	24	3.48	100	0	348.00	121	GCMS
1300Terminal																			
1430Terminal																			
Ferrite Bead	2673000101	FRT	NA	N	N	0	0	2		13	26	0	26	0.16	200	0	32.00	186	GCMS
												Parts Total :			\$97,278				
												Materials Total			\$4,822				
												Program Total			\$102,100				



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Planet - B NMS

Planet B - 080249 0TK63 J0400A-Mech. End View 061-0400 A

RELEASE 29 Feb 96
 UNCLASSIFIED 29 Feb 96
 UNCLASSIFIED 29 Feb 96

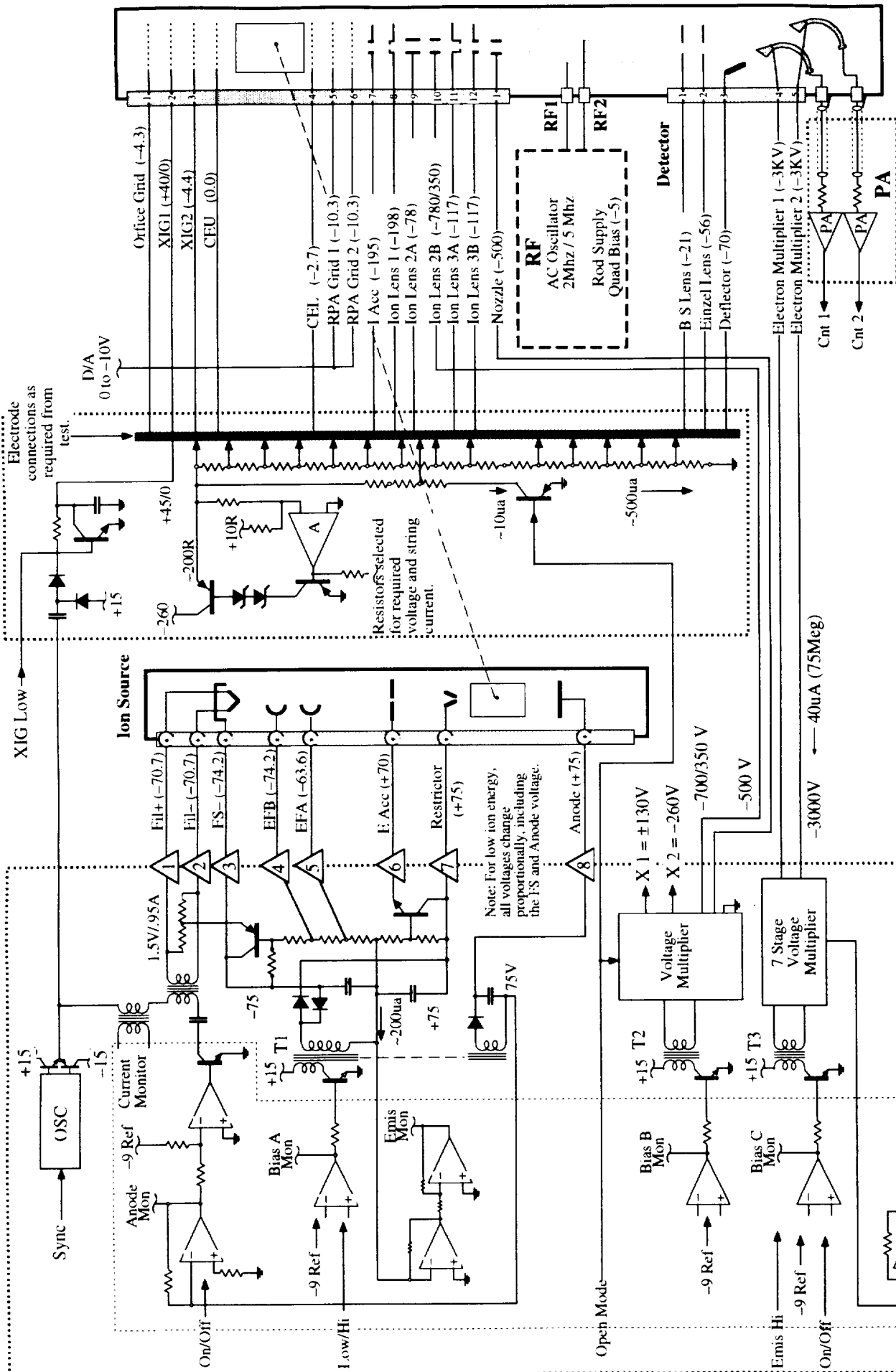
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1 Drawing Page

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Section 2

QP - Quadrupole Electronic Interface



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College of Engineering
University of Michigan

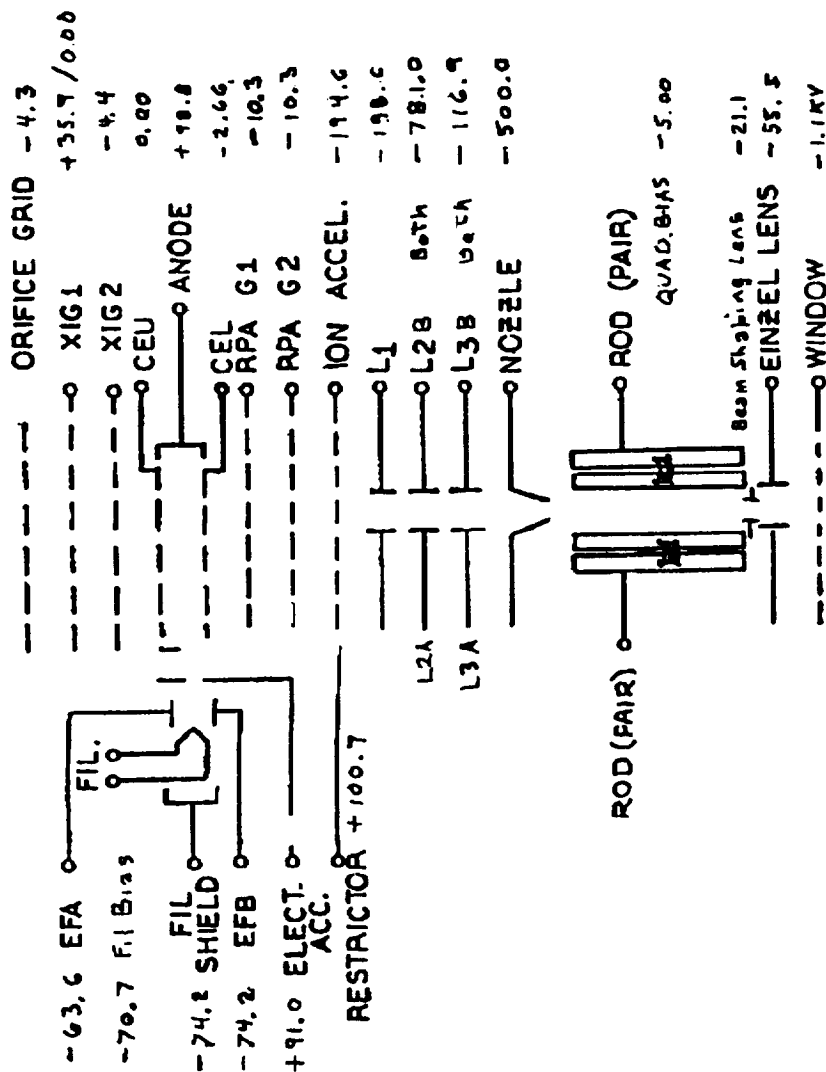
OTK63 NASS-3094330955 16117B-Analyzer Interface 061-0117

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Drawing Page 1

2.02

PV ONMS Electrode Voltages



Received 10-Feb-95 from H. Bentom

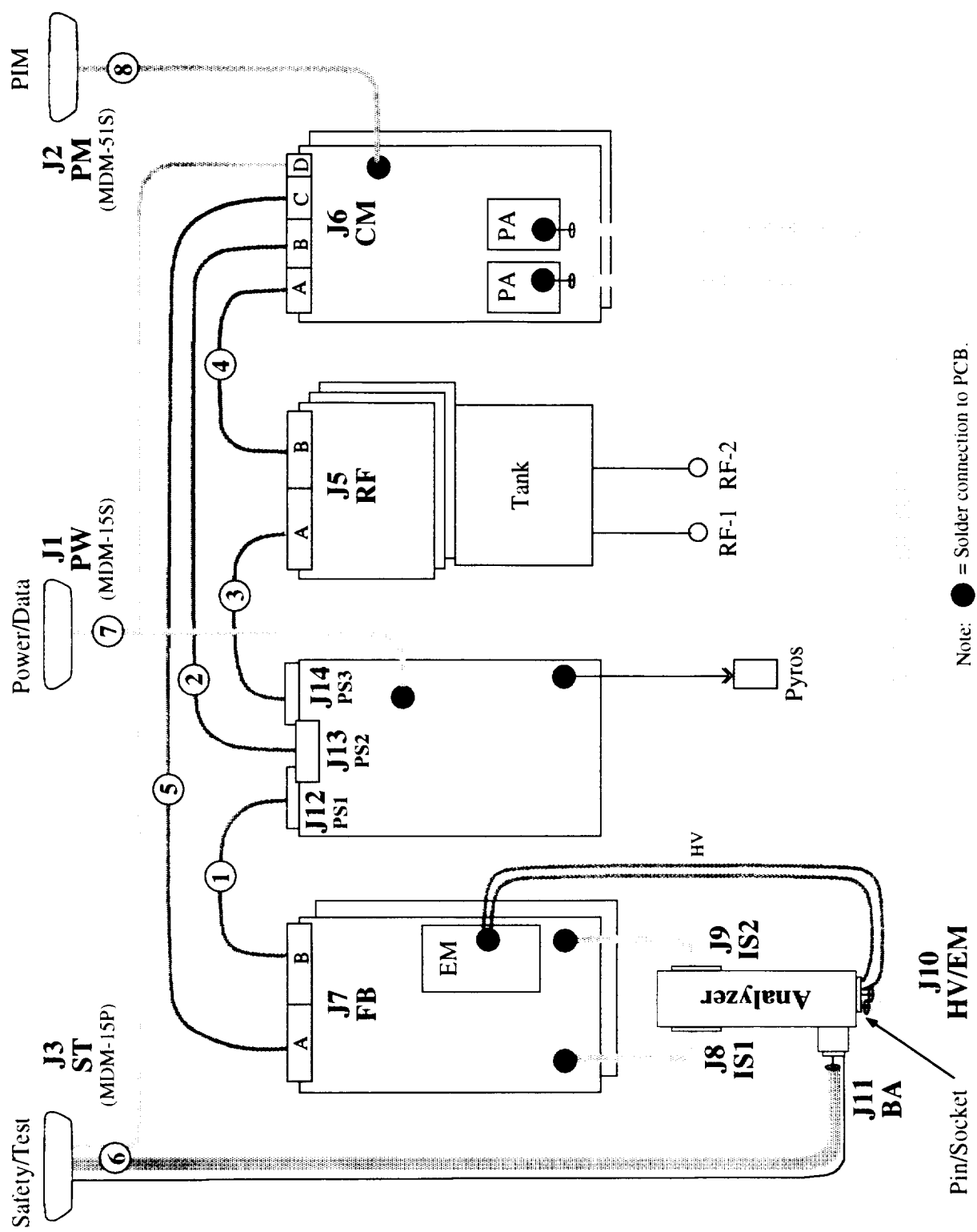
Space Physics Research Laboratory College of Engineering University of Michigan	Quadrupole Electrode Voltages Planet - B
INMS / NAS53094	OTK63
J0115-QP Electrodes	061-0115
DATE: 11-15-95	ENGINEER: M. BENTOM
RELEASE: 12-MAY-95	DRAWN: 11-15-95

Quadrupole Analyzer to Electronics Interface

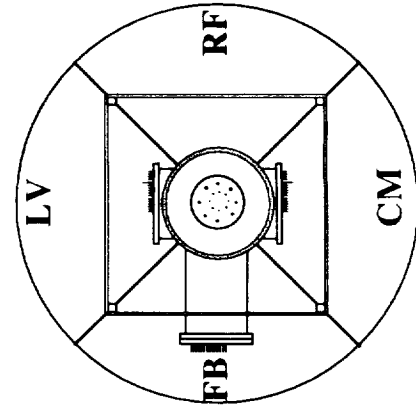
Temp QP Pin # Name	Design Voltage	Function	FB Pin #
1 Orifice Grid	-4.3	Fixed potential, TBS, from -200R divider	TBD
2 XIG1	+45/0	Voltage Doubler from ± 15 AC with added +15 DC, or by XIG LOW cmd to -10.5V	TBD
3 XIG2	-4.4	Fixed potential, TBS, from -200R divider	TBD
4 CEU	0	Ground (from -200R divider)	TBD
5 CEL	-2.7	Fixed potential, TBS, from -200R divider	TBD
6 RPA Grid 1	-10.3/0to10	Fixed at -10(TBS), from -200R divider, or from RPA Value cmd 0 to +10, by Select cmd	TBD
7 RPA Grid 2	-10.3/0to10	Same as above (tied)	TBD
8 I acc	-195	Fixed potential, TBS, from -200R divider	TBD
9 Ion Lens 1	-198	Fixed potential, TBS, from -200R divider	TBD
10 Ion Lens 2A	-78	Fixed potential, TBS, from -200R divider	TBD
11 Ion Lens 2B	-780/350	-780(TBS), or -350(TBS) by Open Mode Cmd	TBD
12 Ion Lens 3A	-117	Fixed potential, TBS, from -200R divider	TBD
13 Ion Lens 3B	-117	Fixed potential, TBS, from -200R divider	TBD
14 Nozzle	-520	4x Voltage multiplier tap from 130 Oscillator - not secondary regulated, and not adjustable.	TBD
15 B S Lens	-21	Fixed potential, TBS, from -200R divider	TBD
16 Einzel Lens	-56	Fixed potential, TBS, from -200R divider	TBD
17 Deflector	-70	Fixed potential, TBS, from -200R divider	TBD
18 EM 1	-3000	-3000 (TBS) volts from 7 stage voltage multiplier. 75 to 100 Meg load.	TBD
19 EM 2	-3000	-3000 (TBS) volts from 7 stage voltage multiplier. 75 to 100 Meg load.	TBD
20 EM Anode 1	0	Coax plug to 50 ohm coax to 3db impedance pad. ~35mV fs (clipping) pulse @14 ns wide.	(CM)
21 EM Anode 1	0	Coax plug to 50 ohm coax to 3db impedance pad. ~35mV fs (clipping) pulse @14 ns wide.	(CM)
22 Fil+	-70/-25	Fil AC at .95A/1.85V rms at 100KHz, referenced to to of fil bias string at -70(TBS), selected by Emis Hi Cmd, OR (NOT D/A!) -25(TBS). Current sum to Emis Mon	TBD
23 Fil-	-70/-25	Same	TBD
24 FS	-72/*	Voltage proportional to Fil. Current sum to Emis Mon	TBD
25 EFA	-74/*	Voltage proportional to Fil. Current sum to Emis Mon	TBD
26 EPB	-74/*	Voltage proportional to Fil. Current sum to Emis Mon	TBD
27 E Acc	70	Voltage proportional to Fil. Current sum to Emis Mon	TBD
28 Restrictor	75	Voltage proportional to Fil. Current sum to Emis Mon	TBD
29 Anode	75	Voltage proportional to Fil. Current sum to Anode Mon and regulates filament temperature.	TBD

*TBS = To Be Selected by resistor selection.

Section 3
HR - Harness, Main & Header



Note: ● = Solder connection to PCB.



End View of Analyzer

Space Physics Research Laboratory College of Engineering University of Michigan		Harness Pictorial	
Planet - B NMS		061-0124	
061-NMS	0TK63	10124A-Harness Pictorial	061-0124
DATE	BY	CHKD	APPD
12-Oct-95	A	12-May-95	A
12-May-95		12-May-95	
DRAWN 12-APR-95 Jm		CHECKED 12-APR-95 Jm	
1		2	
3		4	
5		6	
7		8	
9		10	
11		12	
13		14	
15		16	
17		18	
19		20	
21		22	
23		24	
25		26	
27		28	
29		30	
31		32	
33		34	
35		36	
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79		80	
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85		86	
87		88	
89		90	
91		92	
93		94	
95		96	
97		98	
99		100	

Harness 1	P12	P7B	Arm-2	10	← P6B-9	RF Temp/RF	1	→ CMA-1	Harness 6	J3	CMD
FB 15 Gnd	1	→ P7B-1	Fire-2	11	← P6B-11	-RF AGC	11	→ CMA-2	+5 Inhibit	1	← CMD-1
FB 5 Gnd	2	→ P7B-4	Index Pulse +	12	→ P6B-12	RF Data	14	→ CMA-3	HV Inhibit	2	← CMD-2
FB -15	3	→ P7B-3	Index Pulse -	13	→ P6B-13	RF Clk	15	← CMA-4	Pyro Inhibit	3	← CMD-3
FBAC -15a	4	→ P7B-10	Phase Clock +	14	→ P6B-14	RF AC Enb	2	← CMA-5	BA Anode	4	← B/A-4
FBAC -15b	5	→ P7B-10	Phase Clock -	15	→ P6B-15	RF DC Enb	7	← CMA-6	BA Fil+	5	← B/A-5
FBAC Gnd	6	→ P7B-8	Guard Pulse +	16	→ P6B-16	RF Cal	4	→ CMA-7	BA Fil-	6	← B/A-6
FBAC +15a	7	→ P7B-9	Guard Pulse -	17	→ P6B-17	RF Logic Gnd	6	→ CMA-8	BA Collector	7	← B/A-7
FBAC +15b	8	→ P7B-9	300K Sync/	18	← P6B-7	POR	12	← CMA-9	Col Shield	8	← B/A-8
FB +15	9	→ P7B-2	300K Sync	19	← P6B-6				Therm Press	9	← B/A-9
FB +5a	10	→ P7B-5							Therm Rtn	10	← B/A-10
FB +5b	11	→ P7B-5							Therm Ref	11	← B/A-11
Imon	12	→ P7B-11							Spare	12	
LV Temp	13	→ P7B12							Spare	13	
Imon Drv	14	→ P7B-13							Spare	14	
FB +130	15	→ P7B-6							Spare	15	
FB 130	16	→ P7B-7									

Harness 5 P7A P6C

Emis Hi	3	← CMC-1
Open Mode	12	← CMC-2
AMUX Out	19	← CMC-3
AMUX Add 0	1	← CMC-4
AMUX Add 1	10	← CMC-5
AMUX Add 2	2	← CMC-6
AMUX Add 3	11	← CMC-7
Fil On	13	← CMC-8
EM On	4	← CMC-9
EM Hi	6	← CMC-10
RPA n	14	← CMC-11
100K Clk	25	← CMC-12
Spare	17	← CMC-13
HV On	5	← CMC-14
Multana 1	22	← CMC-15
Multana 2	23	← CMC-16
RF Temp/FB	20	← CMC-17
-RF AGC	21	← CMC-18
XIG Low	7	← CMC-19

Refer to Dwg. 061-0124, Pg. 3.01 for harness numbers and plug identification.

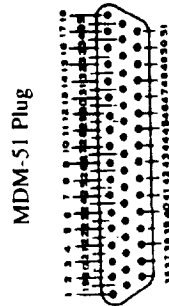
Harness 2 P13 P6B

CM -15	1	→ P6B-3
CM +15	2	→ P6B-2
CM +5a	3	→ P6B-5
CM +5b	4	→ P6B-5
CM 5 Gnda	5	→ P6B-4
CM 5 Gndb	6	→ P6B-4
CM 15 Gnd	7	→ P6B-1
Fire-1	8	← P6B-10
Arm-1	9	← P6B-8

Harness 4 P5B P6A

Table 6.1-1 User Interface Connector pin assign

PIN No	SIGNAL NAME	I/O	NOTE	PIN No	SIGNAL NAME	I/O	NOTE
1				18	REV ASK	0	
19				28	TI DATA ENA	0	
36				45	TI DATA	0	
2	TD15	0		11	REQ	1	
20	TD14	0		29	TI LCK	1	
37	TD13	0		46	TI CLK	1	
3	TD12	0		12			
21	TD11	0		30			
38	TD10	0		47			
4	TD9	0		13	R87	1	
22	TD8	0		31	R06	1	
33	TD7	0		48	R05	1	
5	TD6	0		14	R84	1	
23	TD5	0		32	R03	1	
40	TD4	0		49	R02	1	
6	TD3	0		15	R01	1	
24	TD2	0		33	R00	1	
41	TD1	0		50			
7	TD0	0		16			
25	S1	0		34	+5V	1	
42	S0	0		51			
8	SDPS S7B	0		17	CON(2)	1	
26	WE	0		35			
43	RD	0					
9	MCRRP	0		18			
27	ON CMD	0					
44	OFF CMD	0					



51 Contacts

From Planet-B, Data Handling System, User's Manual, 457 Pg. 6-1, Rev. 5 Scanned 5/9/95

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PIM Plug List

Planet – B NMS

UNIVERSITY OF MINNESOTA	LIBRARY	DATE	061-0136
061-NMS	0TK63	19136-PM Plug list	

2
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Drawing Page 3.03

J1- Power/Data MDM-15S

- 01- Shield Ground
- 02- Index Pulse +
- 03- Phase Clock +
- 04- Guard Pulse +
- 05- Not Used
- 06- Bus-1 (Power +)
- 07- Not Used
- 08- Not Used
- 09- Not Used
- 10- Index Pulse -
- 11- Phase Clock -
- 12- Guard Pulse -
- 13- Not Used
- 14- Com-1 Power Return
- 15- Not Used

Notes:
1- This pin assignment only applies to the Engineering Unit.

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NMS Power Connector List
Planet - B NMS

061-NMS 0TK63 J0150-PowerPlug 061-0150 -
RELEASE 26-May-95
DRAWN 26-May-95 Jm
CHECKED
DATE

P12
TBD
OMNETICS

- ① FB -15 GND
- ② FB -5 GND
- ③ FB -15
- ④ FBAC -15
- ⑤ FBAC -15
- ⑥ FBAC GND
- ⑦ FB AC +15
- ⑧ FB AC +15
- ⑨ FB +15
- ⑩ FB +5
- ⑪ FB +5
- ⑫ IMON
- ⑬ LV TEMP
- ⑭ IMON DRV
- ⑮ FB +130
- ⑯ FB -130

HARNESS 1

PS

FB

- ① TBD
- ② TBD
- ③ TBD
- ④ TBD
- ⑤ TBD
- ⑥ TBD
- ⑦ TBD
- ⑧ TBD
- ⑨ TBD
- ⑩ TBD
- ⑪ TBD
- ⑫ TBD
- ⑬ TBD
- ⑭ TBD
- ⑮ TBD
- ⑯ TBD

J7B
TBD
OMNETICS

PRELIMINARY

Space Physics Research Laboratory College of Engineering University of Michigan		Harness 1		Planet - B NMS		061-0338		A		27-Feb-96 DC RELEASE 18-AUG-95 DRAWN: 26-Jul-95 CHKD: EDC A	
061-NMS		0TK63 J0338A-Harness		2		2/27/1996 12:55 PM		2		Drawing Page 3.05	

P13
TBD
OMNETICS

- ① CM -15
- ② CM +15
- ③ CM +5
- ④ CM +5
- ⑤ CM 5 GND
- ⑥ CM 5 GND
- ⑦ CM 15 GND
- ⑧ FIRE -1
- ⑨ ARM -1
- ⑩ ARM -2
- ⑪ FIRE -2
- ⑫ INDEX PULSE+
- ⑬ INDEX PULSE-
- ⑭ PHASE CLOCK+
- ⑮ PHASE CLOCK-
- ⑯ GUARD PULSE+
- ⑰ GUARD PULSE-
- ⑱ WORK SYNC/
- ⑲ WORK SYNC

PS



CM



HARNES 2

PGB
TBD

TBD

PRELIMINARY

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Harness 2

Planet - B NMS

061-NMS

0TK63

00346A-Harness 2

061-0346

A

1-MAR-96 DC

18 AUG 95

25 JUL 95

ASST/MAINTENANCE

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Drawing Page

3.06

P14
TBD
OMNETICS

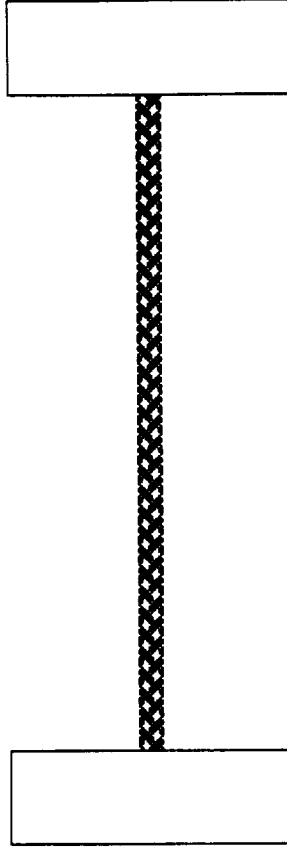
HARNESS 3

J5A
TBD

- ① RF -130
- ② RF +130
- ③ SPARE
- ④ RF +5
- ⑤ RF +5
- ⑥ RF +15
- ⑦ RF +15
- ⑧ RF +15
- ⑨ RF +15
- ⑩ RF -15
- ⑪ RF -15
- ⑫ RF -15
- ⑬ RF -15
- ⑭ RF -5 GND
- ⑮ RF -15 GND

PS

RF



TBD

PRELIMINARY

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Harness 3

Planet - B NMS

061-NMS

0TK63

J0339A-Harness 3

061-0339

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Drawing Page

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3.08

P7A
TBD
OMNETICS

HARNESS 5

P6C

FB

CM



TBD

TBD



PRELIMINARY

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University of Michigan

Harness 5

Planet - B NMS

061-NMS

0TK63

J0345A-Harness 5

061- 0345 / A

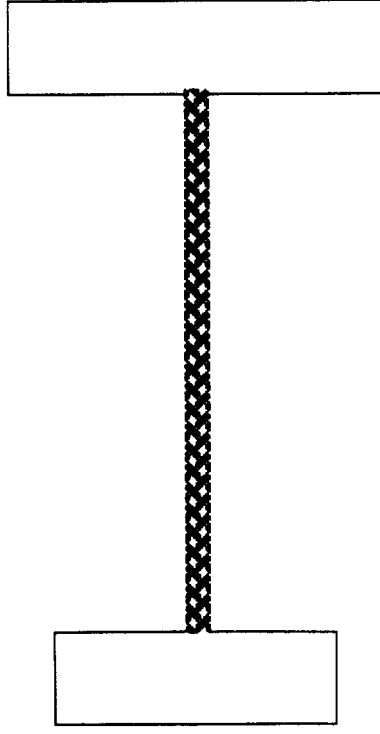
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REV	18-Aug-95	RELEASE
DATE	25-Jul-95	DRAWN
DATE	25-Jul-95	FILED
DATE	25-Jul-95	FILED

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Drawing Page 3.09

HARNESS 6

J3
SAFETY/TEST
INTERFACE



TBD

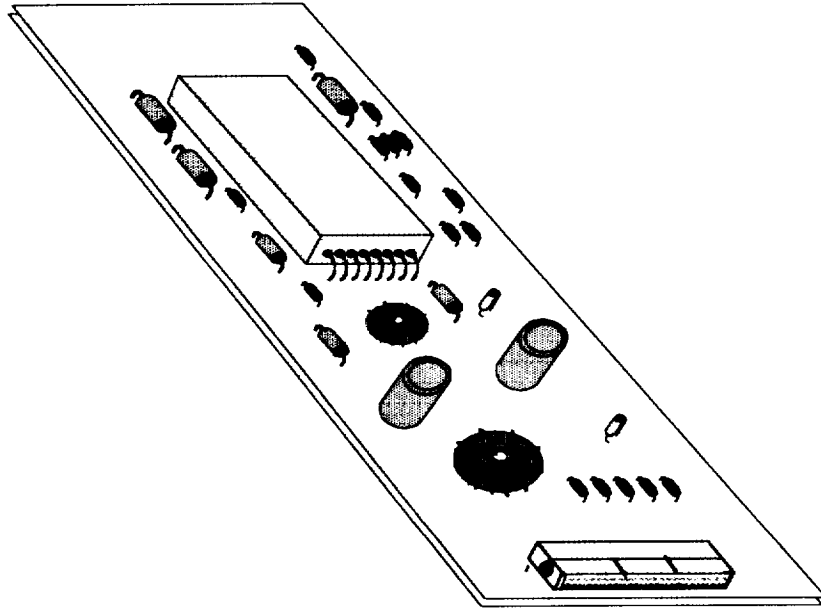
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PRELIMINARY

Space Physics Research Laboratory College of Engineering University of Michigan		Harness 6 Planet - B NMS		REV	DATE
061-NMS	0TK63	J0342A-Harness 6	061-0342	A	1-Mar-96 DC
DRAWN		YK: EMC		26-Jul-95	
CHECKED		Aircraft/Maneuvers			

Section 4

PS - Power Supply Module



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University of Michigan

PS Pictorial
Planet-B NMS

061-NMS 0TK63 J0102-PS Pictorial

061-0102

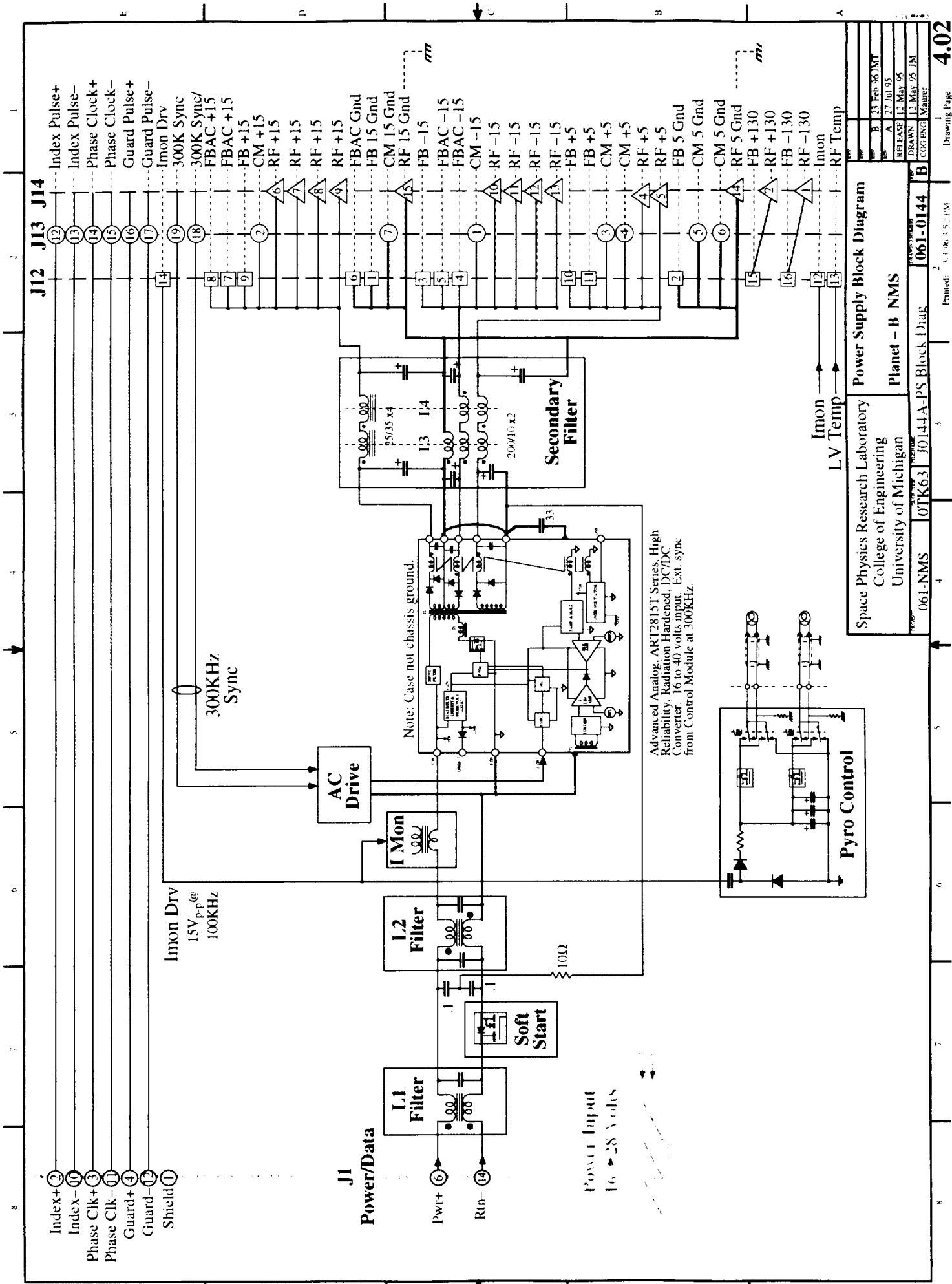
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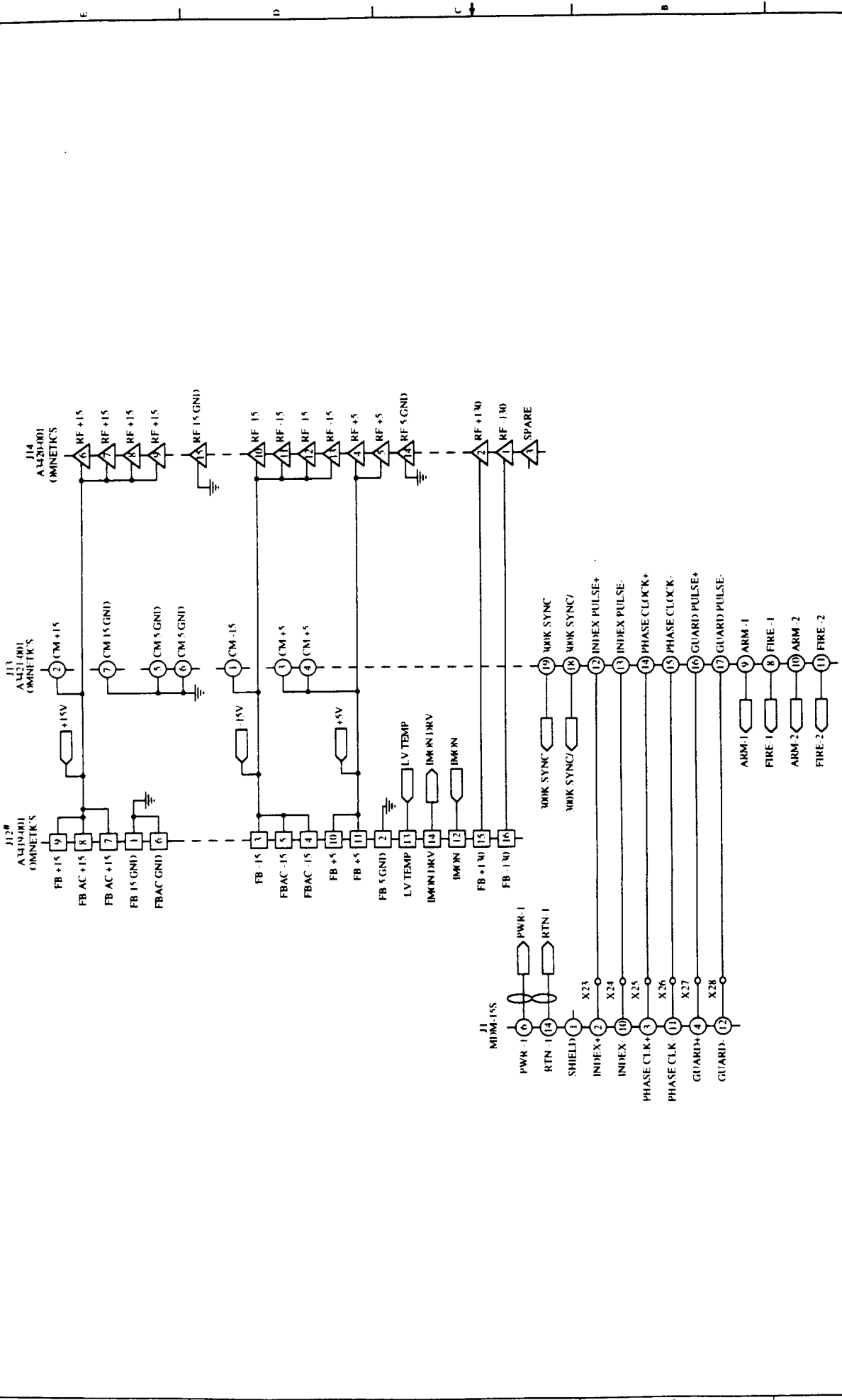
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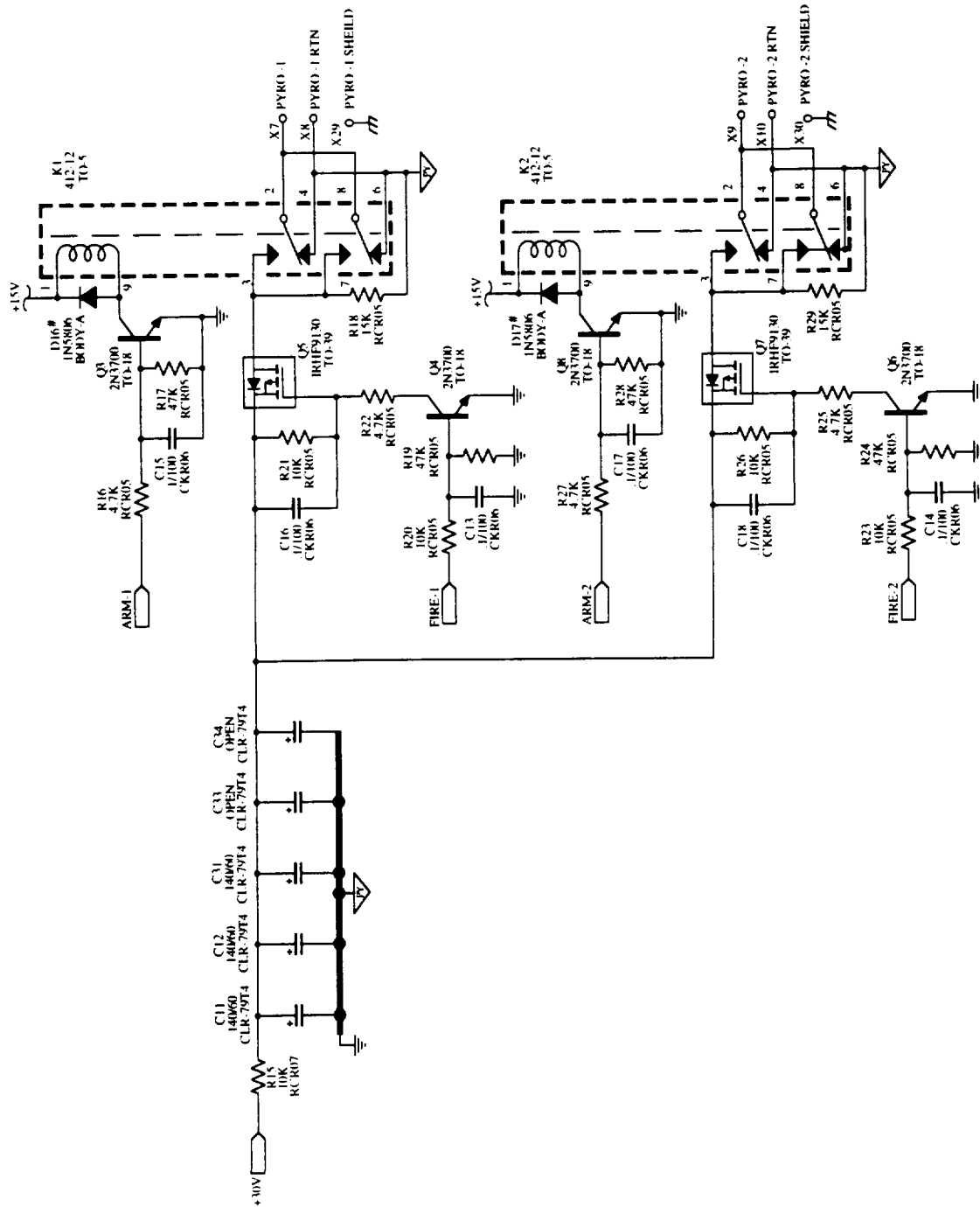
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RELEASE 13-MAY-95
DRAWN 10-MAY-95 JMC
COG ENG ADP/EL





Space Physics Research Laboratory				Schematic Diagram, Pg. 1 of 3			
College of Engineering				Power Supply			
University of Michigan				Planet B - NMS			
PROJECT	Planet B - NMS	DATE	07K63 J0126.1-PS Schematic	REVISION	061-0126	REV	D
DRAWN				CHECKED			
BY				DATE			
D				8-Feb-96 D. Cunningham			
C				12-Oct-95 D. Cunningham			
B11-ASF				10-Aug-95 D. Bay			
B11-ASF				28-Apr-95 D. Bay			
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Space Physics Research Laboratory		Schematic Diagram, Pg. 3 of 3	
College of Engineering		Power Supply	
University of Michigan		Planet B - NMS	
PROJECT	DATE	REVISION	BY
Planet B - NMS	07K63 J0126.3-PS Schematic	061-0126	D
DESIGNED BY	DATE	REVISION	BY
061-0126	28 Apr 95	1	0 Bayl
061-0126	10 Aug 95	2	0 Bayl
061-0126	12 Oct 95	3	0 Bayl
061-0126	8 Feb 96	4	0 Bayl

Ref	Value	Info	Ref	Value	Info
C1	.39/400	HV23	D6	IN5711	DO-35
C2	.01/100	CKR05	D9	IN5305	DO-7
C3	.39/400	HV23	D10	IN4148	DO-35
C4	.1/100	CKR06	D11	IN754A	DO-7
C5	.1/100	CKR06	D12	IN5806	BODY-A
C6	56/75	CLR-79T3	D13	IN5806	BODY-A
C7	.1/100	CKR06	D16	IN5806	BODY-A
C8	.01/100	CKR05	D17	IN5806	BODY-A
C9	.33/50	CKR06			
C10	.1/100	CKR06	FB1	FB2673022401	Ferrite Bead
C11	140/60	CLR-79T4	FB2	FB2673022401	Ferrite Bead
C12	140/60	CLR-79T4	FB3	FB2673022401	Ferrite Bead
C13	.1/100	CKR06	FB4	FB2673022401	Ferrite Bead
C14	.1/100	CKR06	FB5	FB2673022401	Ferrite Bead
C15	.1/100	CKR06	FB6	FB2673022401	Ferrite Bead
C16	.1/100	CKR06	FB7	FB2673022401	Ferrite Bead
C17	.1/100	CKR06	FB8	FB2673022401	Ferrite Bead
C18	.1/100	CKR06	FB9	FB2673022401	Ferrite Bead
C19	.33/50	CKR06	FB10	FB2673022401	Ferrite Bead
C20	.01/100	CKR05			
C21	.01/100	CKR05	H1	ART2815T	ADV.ANALOG
C22	22/35	CSR-21C			
C23	.01/100	CKR05	J1	MDM-15S	
C24	.01/100	CKR05	J12	A3419-001	OMNETICS
C25	22/35	CSR-21C	J13	A3421-001	OMNETICS
C26	.01/100	CKR05	J14	A3420-001	OMNETICS
C27	.01/100	CKR05			
C28	220/10	CSR-21D	K1	412-12	TO-5
C29	.1/100	CKR06	K2	412-12	TO-5
C30	.1/100	CKR06			
C31	140/60	CLR-79T4	L1		061-0129
C32	.01/100	CKR05	L2		061-0130
C33	OPEN	CLR-79T4	L3		061-0132
C34	OPEN	CLR-79T4			
			MA1		061-0131
D1	IN5811	BODY-B			
D2	IN5811	BODY-B	Q1	IRHM254	TO-254AA
D3	IN6325	BODY-A	Q2	2N2369	TO-18
D4	IN5806	BODY-A	Q3	2N3700	TO-18
D5	IN5711	DO-35	Q4	2N3700	TO-18

Ref	Value	Info	Ref	Value	Info
Q5	IRHF9130	TO-39	X4	2031A Useco	Turret
Q6	2N3700	TO-18	X5	2031A Useco	Turret
Q7	IRHF9130	TO-39	X6	2031A Useco	Turret
Q8	2N3700	TO-18	X7	2031A Useco	Turret
R1	100K	RNC50	X8	2031A Useco	Turret
R2	100	RCR05	X9	2031A Useco	Turret
R3	165K	RNC50	X10	2031A Useco	Turret
R4	2.2K	RCR05	X11	2031A Useco	Turret
R5	2.7	RCR07	X12	2031A Useco	Turret
R7	10	RCR07	X13	2031A Useco	Turret
R8	4.7K	RCR05	X14	2031A Useco	Turret
R9	47K	RCR05	X15	2031A Useco	Turret
R10	100	RCR07	X16	2031A Useco	Turret
R11	1.78K	RNC50	X17	2031A Useco	Turret
R12	1K	RCR05	X18	2031A Useco	Turret
R13	30K	YSI	X19	2031A Useco	Turret
R14	100	RCR07	X20	2031A Useco	Turret
R15	10K	RCR07	X21	2031A Useco	Turret
R16	4.7K	RCR05	X22	2031A Useco	Turret
R17	47K	RCR05	X23	2031A Useco	Turret
R18	15K	RCR05	X24	2031A Useco	Turret
R19	47K	RCR05	X25	2031A Useco	Turret
R20	10K	RCR05	X26	2031A Useco	Turret
R21	10K	RCR05	X27	2031A Useco	Turret
R22	4.7K	RCR05	X28	2031A Useco	Turret
R23	10K	RCR05	X29	2031A Useco	Turret
R24	47K	RCR05	X30	2031A Useco	Turret
R25	4.7K	RCR05	X31	2031A Useco	Turret
R26	10K	RCR05			
R27	4.7K	RCR05			
R28	47K	RCR05			
R29	15K	RCR05			
R30	33K	RCR05			
T1	M21038/27-05				
X1	2031A Useco	Turret			
X2	2031A Useco	Turret			
X3	2031A Useco	Turret			

Extracted Thursday, February 08, 1996 02:28 PM

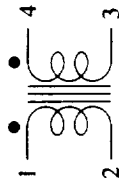
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	FB10	1	J14	12	C26	1	J1	10	
	J12	10	J14	13	C28	2	X24	2	
	J12	11	15 RTN			2	J13	13	
	J13	3	C19	1	C30	2	X24	1	
	J13	4	C20	1	C32	2	J13	13	
	J14	4	C21	2	D5	2	X24	1	
	J14	5	C24	1	FB6	1	J1	3	
	J14	5	C27	2	J12	2	X25	2	
+15V	C22	1	FB5	2	J12	6	J1	3	N:6
	C32	1	X19	1	J13	5	X25	2	
	D12	2	X22	1	J13	6	J13	14	
	D16	1	300K SYNC			7	X25	1	
	D17	1	J13	19	J14	14	J1	11	
	FB4	1	R10	1	J14	15	X26	2	
	J12	7	300K SYNC/			2	J13	15	
	J12	8	J13	18	R30	1	X26	1	
	J12	9	R14	1	X2	2	J1	11	
	J13	2	ARM-1		IMON		X26	1	
+30V	J14	6	J13	9	C10	2	J13	16	N:10
	J14	7	R16	1	J12	12	X27	1	
	J14	8	ARM-2		R12	1	J13	16	
	J14	9	J13	10	IMON DRV		X27	1	
	K1	1	R27	1	C7	2	J1	4	
	K2	1	CHAS		C29	1	X27	2	
	C30	1	H1	65	J12	14	J1	12	
	D13	1	R5	2	LV TEMP			2	
	R15	1	X29	2	J12	13	X28	2	
	R30	2	X30	2	R13	1	J13	17	
-15V			X31	1	N:1			1	N:15
			FIRE-1		J12	16	X28	1	
			J13	8	J14	1	C7	1	
			R20	1	N:2			4	
			FIRE-2		J12	15	MA1	4	
			J13	11	J14	2	D5	1	
			R23	1	N:3			2	
			GND		J1	2	D6	2	
			C10	1	X23	2	MA1	3	
			C22	2	N:4			3	

N:18	C29 D12 D13	2 1 2	N:27	Q1 C4 C5 R5	1 2 1 1	N:36	R10 T1	2 1	N:47	FB5 L3	1 8
N:19	C19 C20 H1	2 2 95	N:28	D4 R7	2 1	N:37	R11 X1	1 2	N:48	FB3 L3	1 6
N:20	C8 C9 H1	2 2 48	N:29	C6 D9 MA1 R7 X13	1 2 2 2 1	N:38	H1 X31	12 2	N:49	FB7 L3	1 5
N:21	D6 R12 X1	1 2 1	N:30	MA1 X15	1 2	N:39	R11 X2	2 1	N:50	FB4 L3	2 2
N:22	C3 C4 D4 FB1 X11	1 1 1 1 2	N:31	C1 D2 D3 Q1 R3 X6	2 2 2 3 2 1	N:40	C27 FB9 X21	1 2 1	N:51	FB9 L3	1 7
N:23	C1 R2 R3	1 1 1	N:32	R8 T1	1 4	N:41	R14 T1	2 2	N:52	FB8 L3	2 1
N:24	D3 Q1 R1 R2 R4	1 2 2 2 1	N:33	D10 Q2 R8 R9	1 2 2 1	N:42	C24 FB7 X20	2 2 1	N:53	FB10 L3	2 3
N:25	D1 D2	2 1	N:34	H1 X15	1 1	N:43	C21 FB3 X18	1 2 1	N:54	L1 X3	1 1
N:26	C2 R4 C2 FB2	1 2 2 2	N:35	D9 D11 Q2 X17	1 1 1 2	N:44	D1 FB1 R1 X5	1 2 1 1	N:55	L1 X4	4 1
						N:45	C3 C5 FB2 X12	2 2 1 2	N:56	L1 X5	2 2
						N:46	FB6 L3	2 4	N:57	L1 X6	3 2
									N:58	L2 X11	1 1
									N:59	L2 X12	4 1
									N:60		

N:61	L2 3 X14 2	N:73	Q6 2 R23 2 R24 1	N:80	R19 1 R20 2	R28 2 R29 1 R29 2 X8 2 X10 2
N:62	L2 2 X13 2		CI7 1 Q8 2 R27 2 R28 1		CI5 1 Q3 2 R16 2 R17 1	PWR RTN C6 2 C8 1 C9 1 D10 2 D11 2 Q2 3 R9 2 T1 3 X14 1 X16 2
N:63	HI 2 X16 1	N:74	D17 2 K2 9 Q8 1	N:81	D16 2 K1 9 Q3 1	
N:64	HI 4 X17 1					
N:65	HI 11 X18 2	N:75	K2 2 K2 8 X9 2	N:82	K1 3 K1 7 Q5 1 R18 1	
N:66	HI 10 X19 2	N:76	CI1 1 CI2 1 CI6 1 CI8 1 C31 1 C33 1 C34 1 Q5 3 Q7 3 R15 2 R21 1 R26 1	N:83	CI1 2 CI2 2 CI3 2 CI4 2 CI5 2 CI7 2 C31 2 C33 2 C34 2 K1 4 K1 6 K2 3 K2 4 K2 6 Q3 3 Q4 3 Q6 3 Q7 1 Q8 3 R17 2 R18 2 R19 2 R24 2	PWR-1 J1 6 X3 2 RTN-1 J1 14 X4 2
N:67	HI 13 X21 2					
N:68	HI 14 X22 2					
N:69	K1 2 K1 8 X7 2	N:77	CI6 2 Q5 2 R21 2 R22 1 Q4 1 R22 2			Extracted Thursday February 8, 1996 10:54
N:70	CI8 2 Q7 2 R25 1 R26 2					
N:71	Q6 1 R25 2	N:78	Q4 1 R22 2			
N:72	CI4 1	N:79	CI3 1 Q4 2			

LIST OF MATERIALS

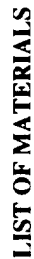
Item	QTY	P/N	Description
1	1	204T250-3C81	Philips/ Ferroxcube Toroid
2	AR	B2222512	#22 AWG Bifilar Magnet Wire
3	AR	3M Scotchcast 235	Encapsulant



NOTES:

- 1) Wind one full layer of #22 AWG bifilar wire on 204T250-3C81 core (~ 14 turns). DO NOT cross finish leads over start leads.
- 2) • denotes phasing.
- 3) Refer to Special Assembly Procedure 061-0389 for Flight Hardware Assembly.

Space Physics Research Laboratory College of Engineering University of Michigan	L1 Assembly Detail Power Supply Planet - B NMS	061-NMS	0TK63	J0129B-PS L1 Ass'y Detail	061-0129	B	26-Feb-95 JMT 12-May-95 O. BAY 08-May-95 O. BAY COG ENG S. Barfield K. A. West	4.20
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Item	QTY	P/N	Description
1	1	A-390134-2	Arnold MPP Core
2	AR	B2222512	#22AWG Bifilar Magnet Wire
3	AR	3M Scotchcast 235	Encapsulant

- 1) Wind 30 turns of #22 AWG bifilar wire on A-390134-2 core. DO NOT cross finish leads over start leads.

2) • denotes phasing.

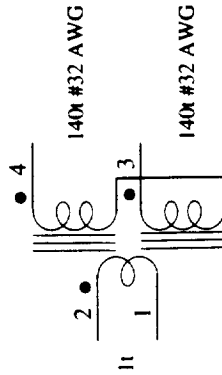
33) Refer to Special Assembly Procedure 061-0390 for Flight Hardware Assembly.

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LIST OF MATERIALS

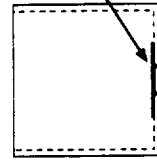
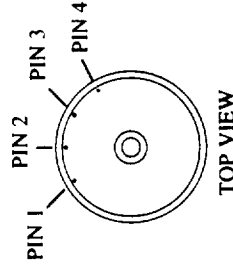
Item	QTY	P/N	Description
1	2	50402-1E	Magnetics Toroid
2	AR	HAPT-32	#32AWG Magnet Wire
3	AR	3M Scotchcast 235	Encapsulant
4	1	061-0131-1	Robison Potting Shell, Modified
5	1	061-0131-2	G10 Insulator
6	1	061-0131-3	G10 Spacer
7	1		Screw, #2-56 x 3/16" Philips 100° FH
8	AR	QPL	#22A WG Solid Bus Wire, Tinned
9	AR	3M 1838B/A	Scotchweld Epoxy
10	AR	Rapid Seal	Latex Sealant
11	AR	QQ-S-571E	Solder
12	AR	EN-11A/B	Conathane Potting

WINDING DETAIL:



- 1) Wind 140 turns of #32 AWG bifilar wire, stack wound on each 50402-1E Magnetics core.
- 2) 2 cores are stacked with common 1 turn primary. Stack up so M and H supplies add.
- 3) Refer to Special Assembly Procedure 061-0391 for Flight Hardware Assembly.

ASSEMBLY DETAIL:



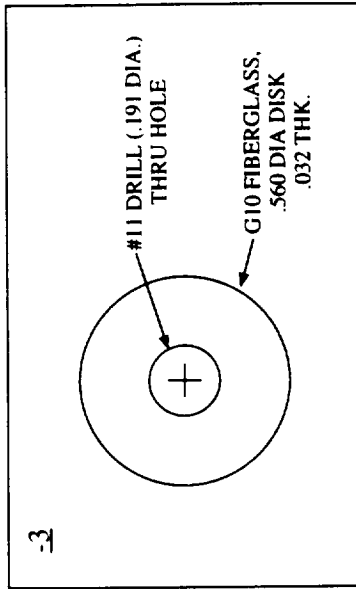
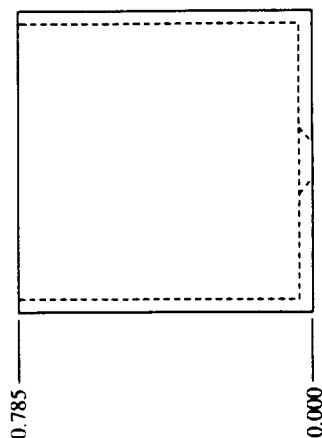
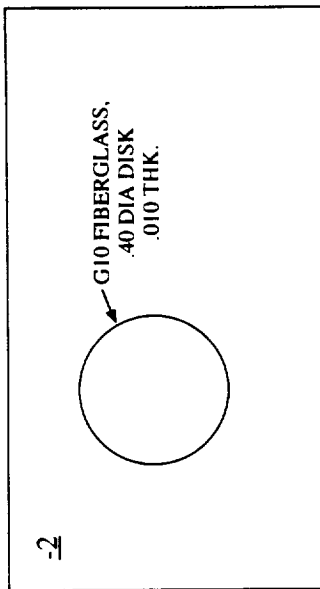
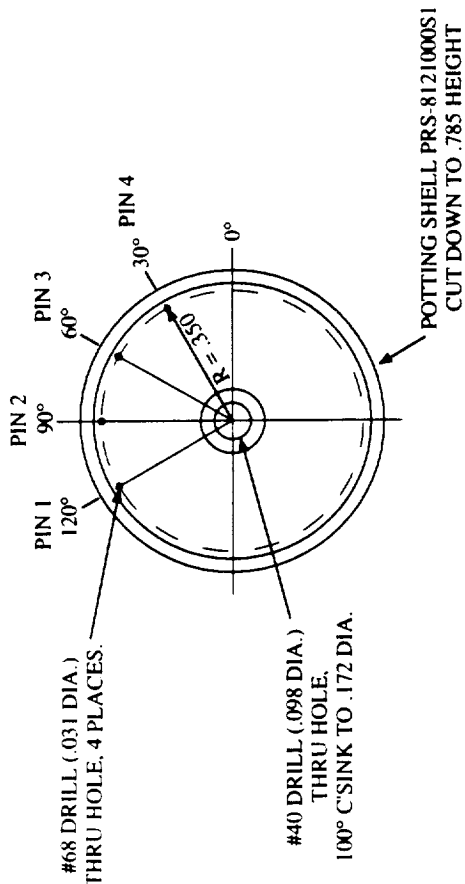
Bond #2-56 screw with 1838B/A.
Bond 061-0131-2 Insulator over screw head.

Bond 061-0131-3 Spacer on bottom of shell with 1838B/A. IMPORTANT: Center on screw.

Space Physics Research Laboratory College of Engineering University of Michigan	MA1 Ass'y Detail, Page 1 of 2 Power Supply Planet - B NMS	REV D 26 Feb 95 C 16 Oct 95 B 05 Jul 95 O. Bayl RELEASE 12 May 95 O. Bayl DRAWN 06 Mar 95 O. Bayl CHECKED S. Ballew K. Amet
061-NMS	0TK63 J0131.1-PS MA1	061-0131.1
Printed: 02/26/96 11:11	Page 1	Drawing Page 4.22

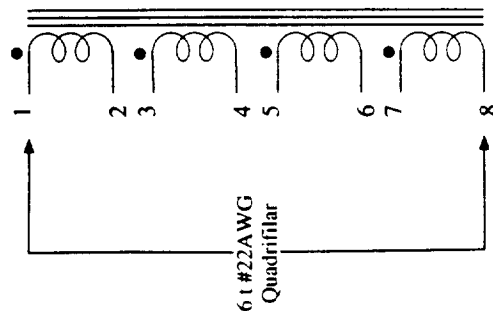
LIST OF MATERIALS

Item	QTY	P/N	Description
1	1	PRS-8121000S1	Robison Potting Shell
2	1		G10, .40 Dia., .010 Thick
3	1		G10, .560 Dia., .032 Thick
4	1		#11 Drill
5	1		#40 Drill
6	1		#68 Drill



Space Physics Research Laboratory College of Engineering University of Michigan		MA1 Ass'y Detail, Page 2 of 2 Power Supply Planet - B NMS		REV D 26-Feb-96 REV C 16-Oct-95 REV B 05-Jul-95 O. Bayl RELEASE 12-May-95 O. Bayl DRAWN 08-Mar-95 O. Bayl CHECKED S. Buel/K. Amst	
061-NMS	0TK63	J0131.2-PS MA1	061-0131.2D	Printed 02/26/96 11:31 Drawing Page 4.23	

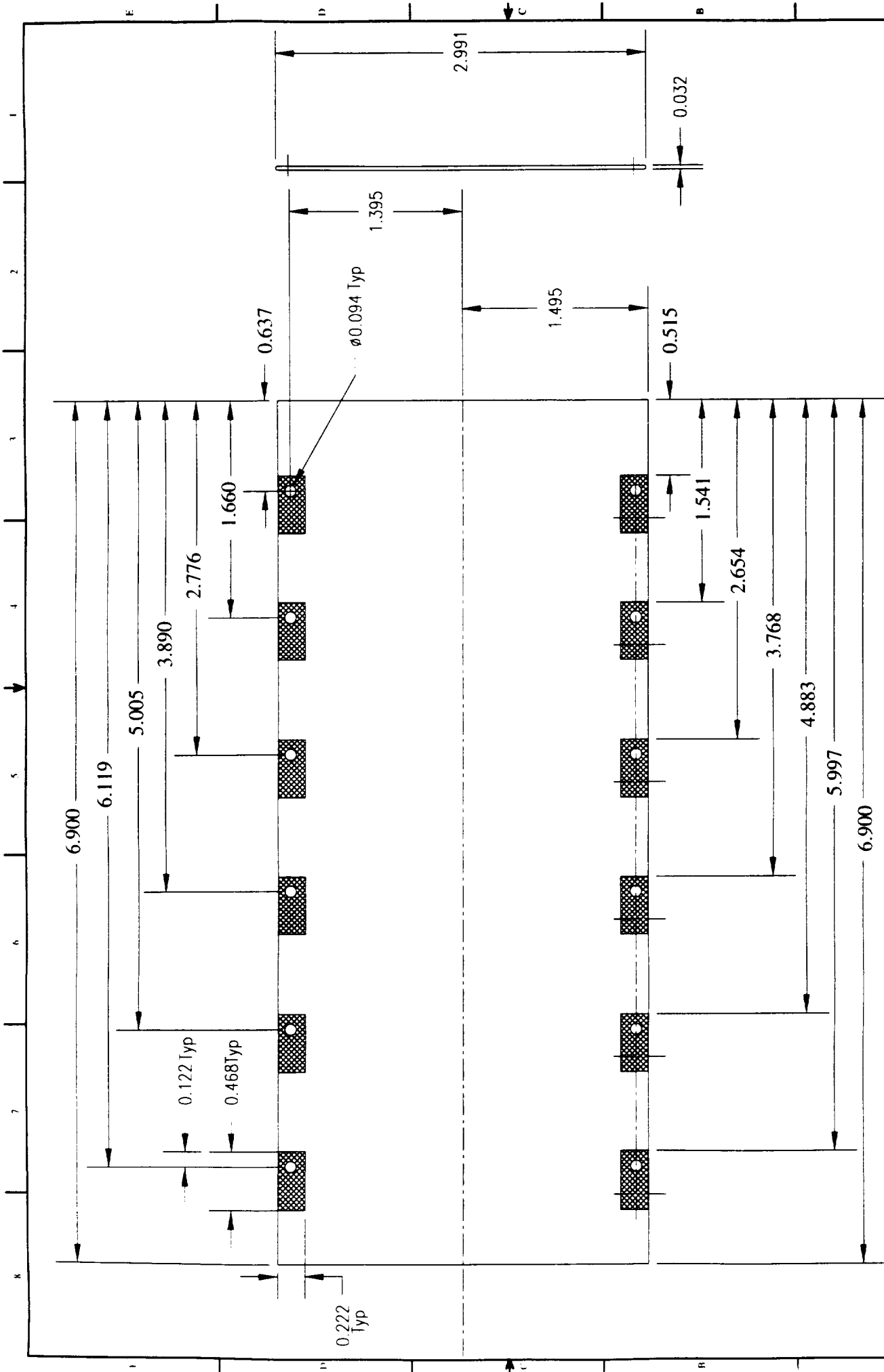
Item	QTY	P/N	Description
1	1	204T250-3C81	Ferrocube Core
2	AR	HAPT-22	#22 AWG Magnet Wire
3	AR	3M Scotchcast 235	Encapsulant



- 1) Wind 6 turns of #22 AWG Quadrifilar wire (single layer) on Ferroxcube core 204T250-3C81. DO NOT cross finish leads over start leads.

2) • denotes phasing.

3) Refer to Special Assembly Procedure 061-0389 for Flight Hardware Assembly.



TYPICAL LAMINATION SEQUENCE
NOT TO SCALE

2 OZ LAYER 4 SOLDER SIDE

GIN C MATERIAL

1 OZ LAYER 3 SIGNAL

PREPREG B STAGE

GIN C MATERIAL

1 OZ LAYER 2 GROUND

PREPREG B STAGE

2 OZ LAYER 1 COMPONENT SIDE

GIN C MATERIAL

TOTAL HOLES: 320

[illegible]

2 OZ.
LAYER 4 SOLDER SIDE

TYPICAL LAMINATION SEQUENCE
NOT TO SCALE

VIEW FROM LAYER 1 (TOP)

100° C SINK 0214 MAJOR DIA.
FOR #2-56 FH SCREW FAR SIDE

100" C'SINK 0142 MAJOR DIA.
FOR 80-80 FH SCREW 6 PLACES

— 0069

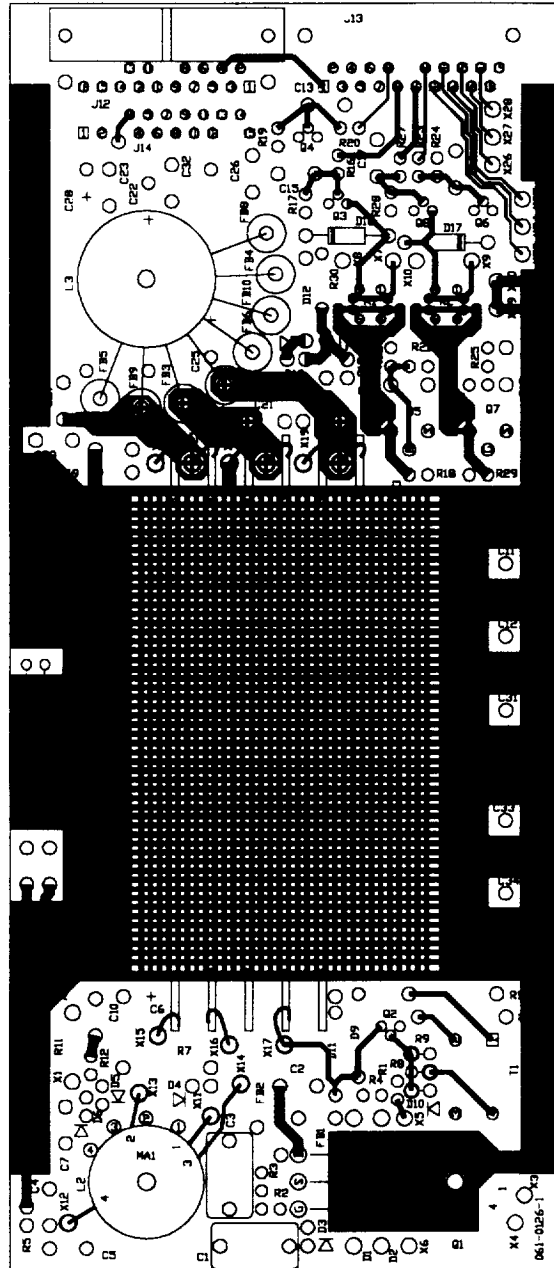
UNLESS OTHERWISE SPECIFIED TOLERANCE ARE:
FRACTIONAL DIMENSIONS ± 0.005
DIMENSIONS IN PARENTHESES ± 0.010
ANGULAR DIMENSIONS ± 30 MIN.

[illegible]

1-03740	1/15
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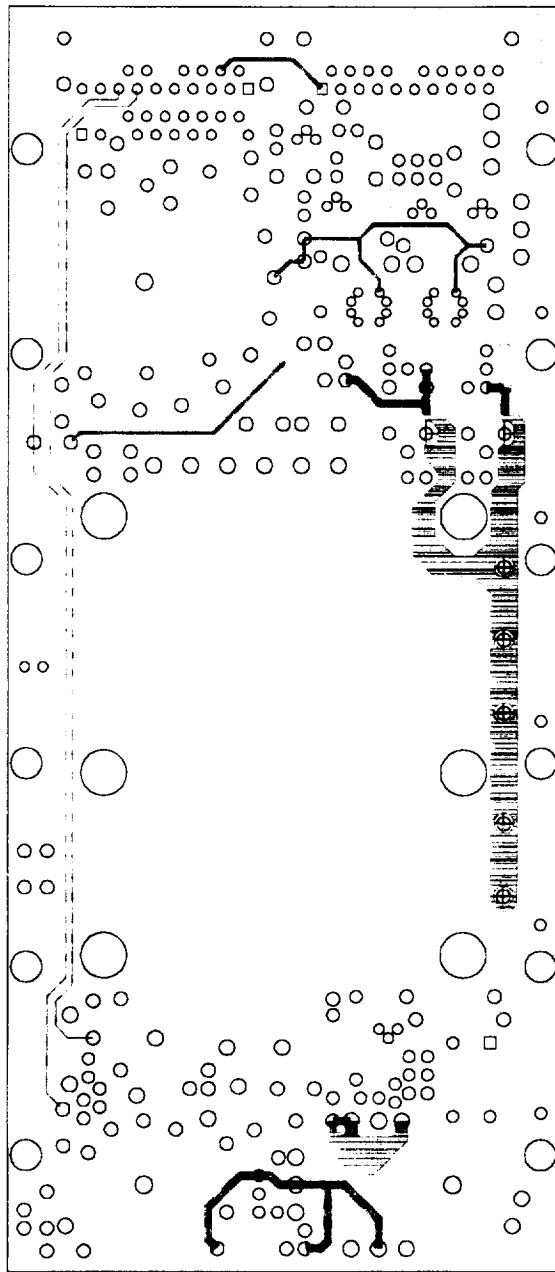
INSTRUMENT	
MATERIAL:	
100	
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60	
40	
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RELEASE: Not Released	
DRAWN: 7-Feb-96 LDISI	
CURING: Amvet / Baicel	

DRAWING PAGE: 4.51



PRELIMINARY

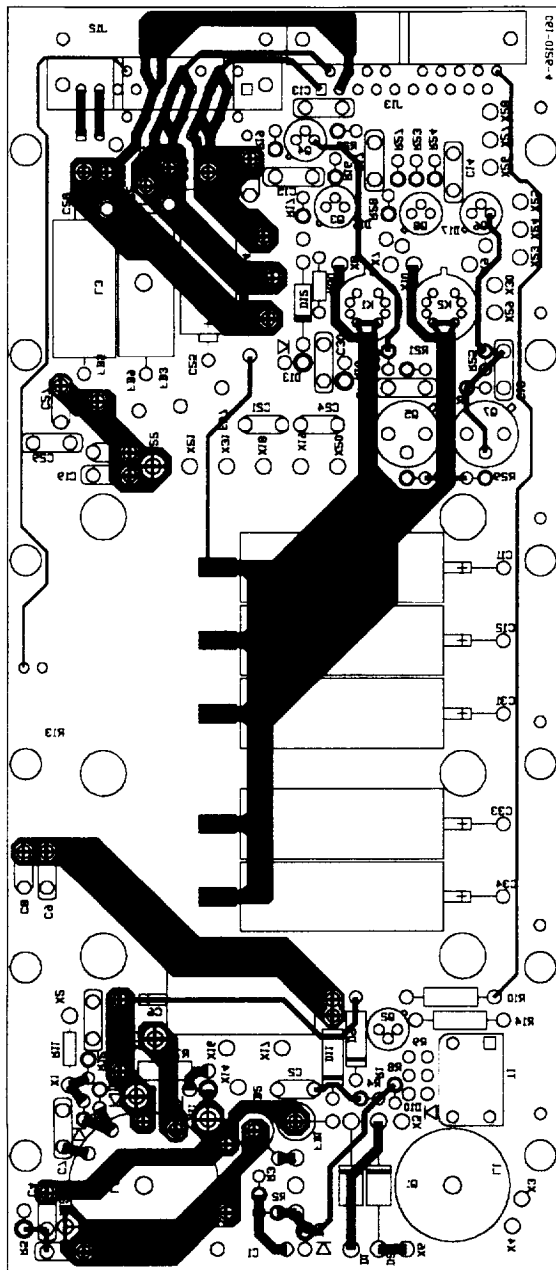
						061-0126				Printed Circuit Board							
L.H.		R.H.		L.H.		R.H.		PART NO		NAME		SIZE		DESCRIPTION			
NO REQD				DASH NO													
Space Physics Research Laboratory College of Engineering University of Michigan												PCB Layer 1 PS Planet B - NMS					



PRELIMINARY

L.H. R.H.		NO REQD.		PART NO.		DASH NO.		NAME		SIZE		DESCRIPTION	
				061-0126				Printed Circuit Board					
Space Physics Research Laboratory				College of Engineering				University of Michigan				PCB Layer 3	
												PS	
												Planet B - NMS	
Planet B - 080249				J0374.3A-PS PCB Lvt. 3				061-0274.3				A	
UNLESS OTHERWISE SPECIFIED TOLERANCES ARE:				DIM. ENDING .000				FRACTIONS				DRAWN 1 Feb 66 EDS	
												RELEASE Not Released	
												COO ENG (A)ELU/Barel	

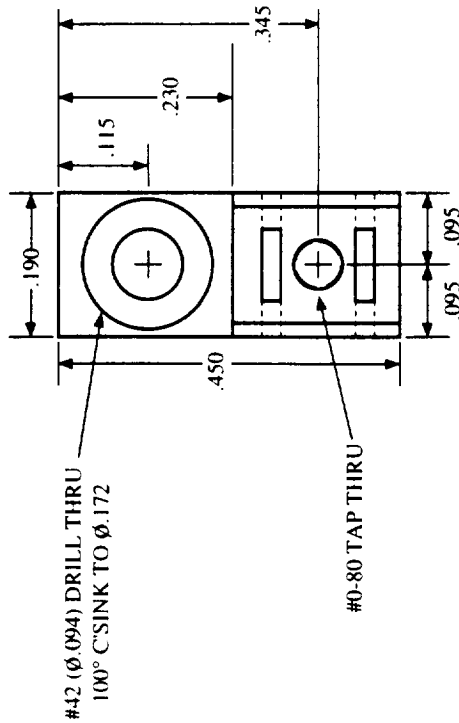
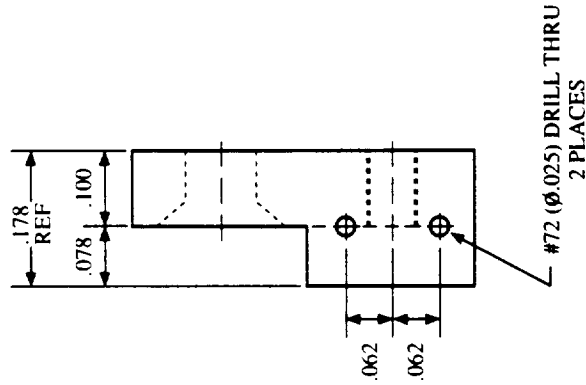
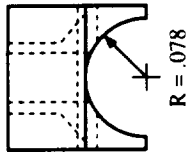
UNLESS OTHERWISE SPECIFIED TOLERANCES ARE:
 DIM. ENDING .000
 DIM. ENDING .000



PRELIMINARY

Printed Circuit Board				SIZE		DESCRIPTION	
L.H.	R.H.	PART NO.	DASH NO.	MATERIAL			
NO REQD							
Space Physics Research Laboratory				PCB Layer 4			
College of Engineering				PS			
University of Michigan				Planet B - NMS			
Planet B - 080249 0TK63 J0374 3A PS PCB Lvt 4				061-0374.4 A		RELEASE Not Released	
						DRAWN 7 Feb 56 EDSI	
						CHECKED ADH/BJE	

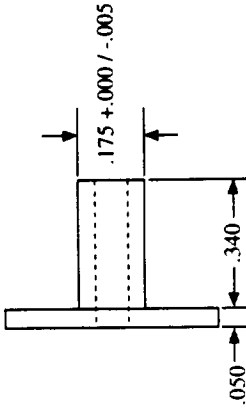
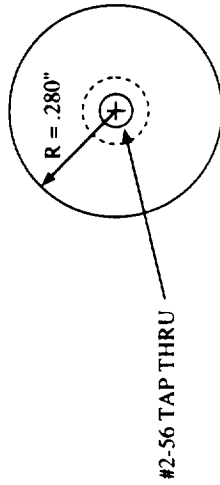
UNLESS OTHERWISE SPECIFIED TOLERANCES ARE:
 DIM. ENDING .00 ±.030 ANGULAR DIM
 DIM. ENDING .000 ±.005 ±.10 MIN.



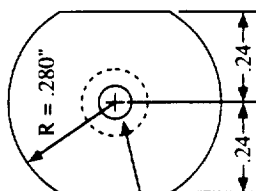
NOTE: BREAK ALL CORNERS

QTY	PART NO.	NAME	SIZE	DESCRIPTION
6	061-0376	UMBILICAL SUPPORT	.178 x .19 x .45	WHITE DELRIN
Space Physics Research Laboratory				
College of Engineering				
University of Michigan				
UMBILICAL SUPPORT				
PS Assembly				
Planet B - NMS				
Planet B - 080249 10TK63 J0376-PS Umb. Support				
061-0376				
RELEASE 1 Feb 96				
DATE 11 Jan 96 BY K. Amet				
CYCLE Amet / Baile				

UNLESS OTHERWISE SPECIFIED TOLERANCES ARE:
DIM FINISHING .001 ±.005
DIM FINISHING .001 ±.005
ANGULAR DIM ±.01 MIN



2		061-0388				INDUCTOR CAP		.280 Dia x .390		WHITE DELRIN	
I. H.		R. H.		I. H.		R. H.		SIZE		DESCRIPTION	
NO REQD		PART NO.		NAME		MATERIAL					
Space Physics Research Laboratory				L1, L3 Cap							
College of Engineering				Power Supply							
University of Michigan				Planet B - NMS							
Planet B - 080249				PART NO.				DRAWING NO.			
061-0388				061-0388				061-0388			
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UNLESS OTHERWISE SPECIFIED TOLERANCES ARE:
DIM. FINISH .00 ±.030
DIM. FINISH .000 ±.005
ANGULAR DIM.
±30 MIN.

4.58

Section 5

FB - Fil/Bias & HV Module

To IS 2 Header

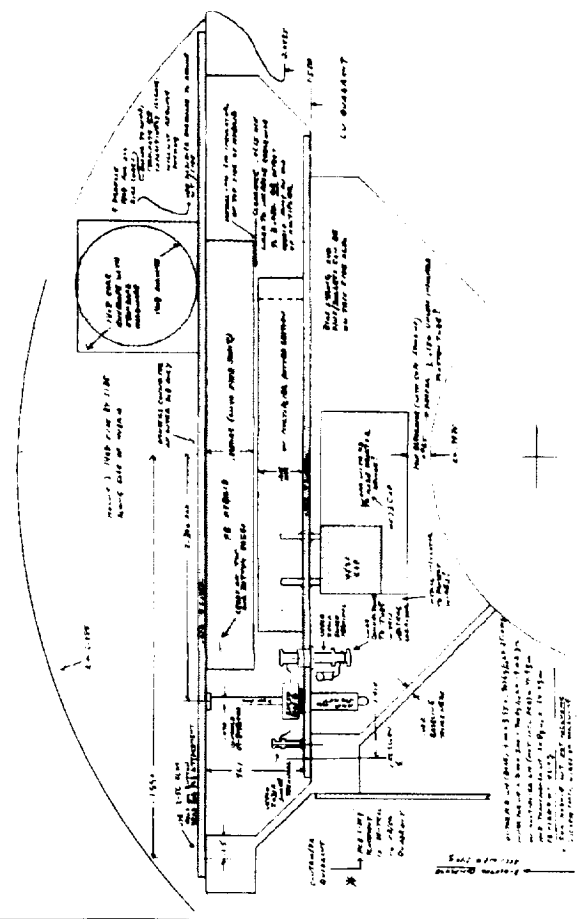
To IS 1 Header

1065 FB Hybrid

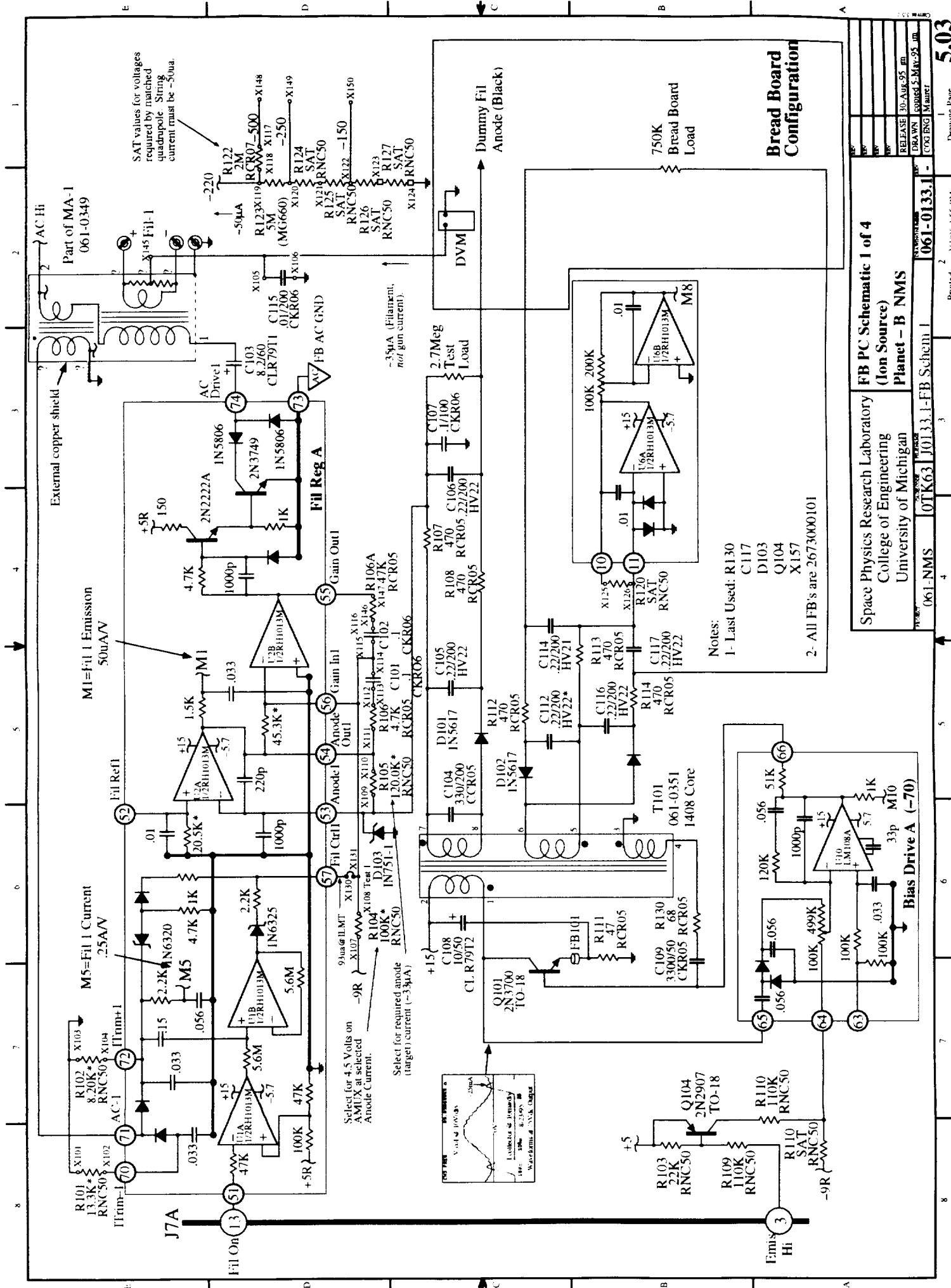
Ion Source Trim Resistors

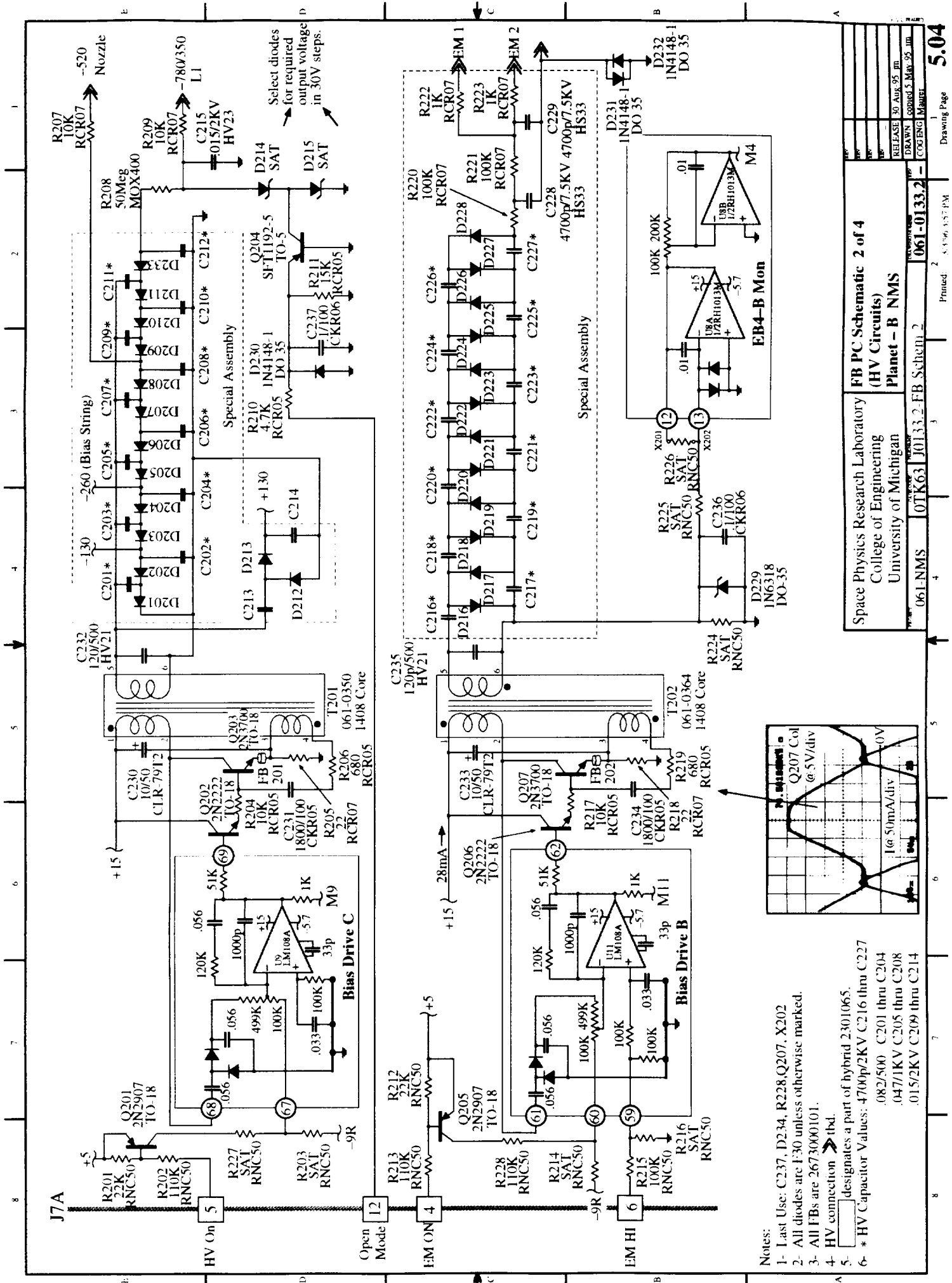
EM HV Voltage Multiplier

HV to Detector Header



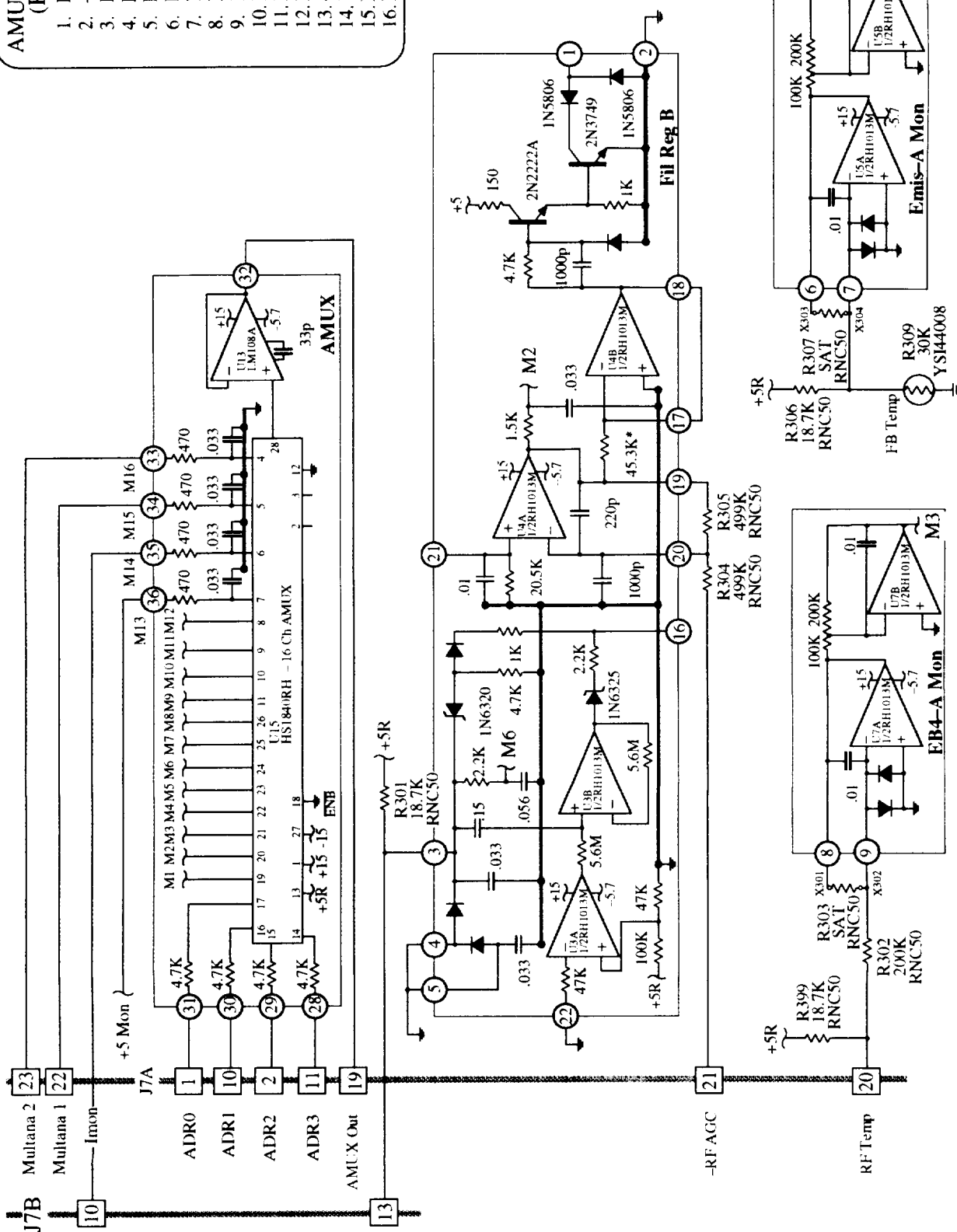
Space Physics Research Laboratory College of Engineering University of Michigan		FB Pictorial Planet-B NMS		061-NMS 10TK63 J0103A-FB Pictorial		061-0103 A		Printed 2 4-3096 2 X 1 PM		Drawing Page 5.01	
A 11-AUG-95 JM		RELEASE 17-MAY-95 JM		DRAWN 10-MAY-95 JM		COO ENG 1 MURET					

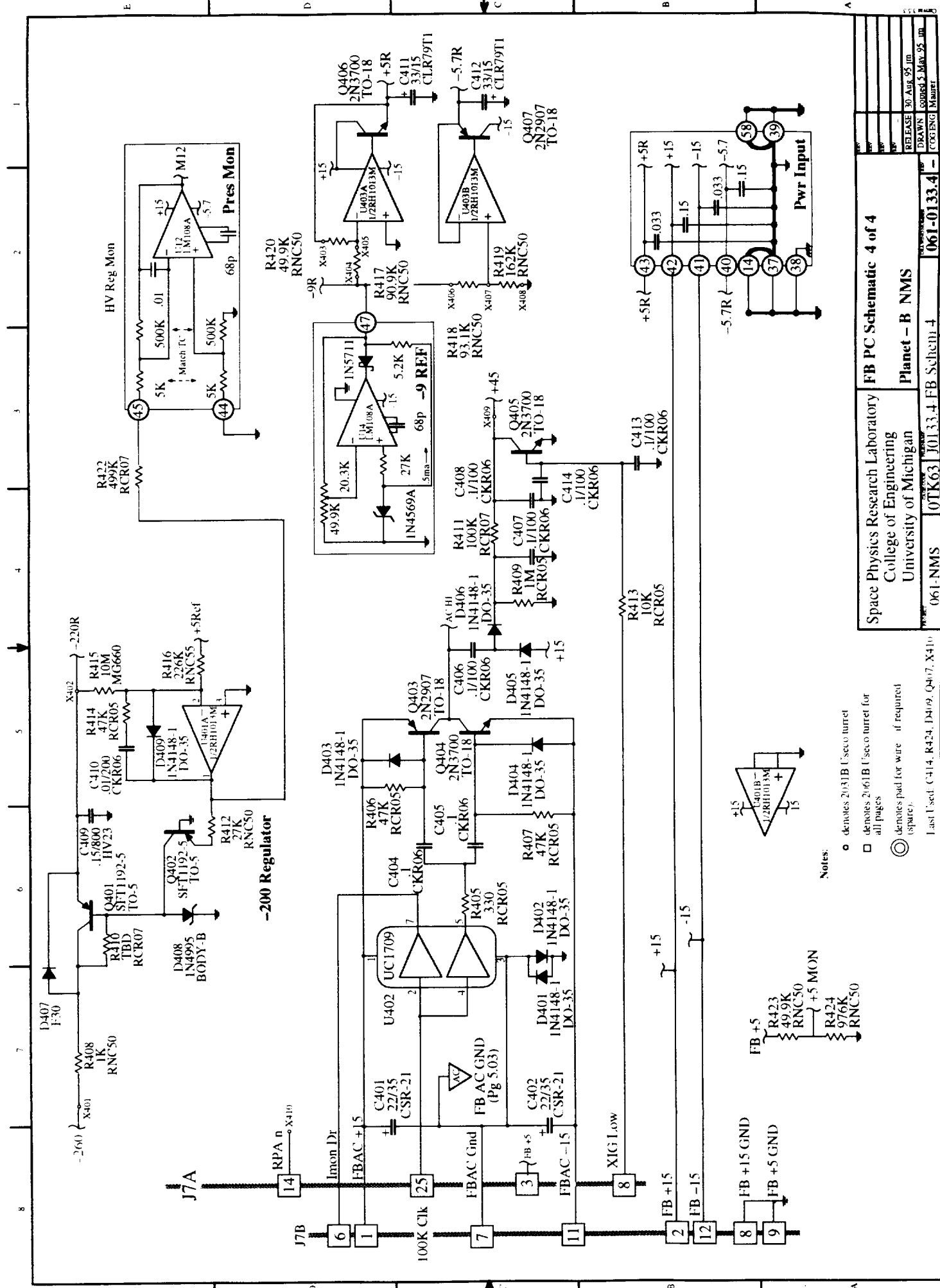




AMUX Assignment (Reference)

1. Fil Anode
2. -RF AGC*
3. RF Temp
4. EM Mon
5. Fil I
6. LVPS Temp
7. F/B Temp
8. Fil Emus
9. Bias C Mon
10. Bias A Mon
11. Bias B Mon
12. HV Reg Mon
13. +5V mon
14. I Mon
15. Multana 1
16. Multana 2





- Notes:
- denotes 2N311B Usecu turret
 - denotes 2N611B Usecu turret for all pages
 - ⊙ denotes pad for wire if required (spare)

Ref	Value	Comment	Part Number	Mfg	Ven	PO	Cost	Date Code	S/N
C101	.1/100	CKR06							
C102	.1/100	CKR06							
C103	8.2/60	CLR79T1							
C104	330/200	CCR05							
C105	22/200	HV22							
C106	22/200	HV22							
C107	.1/100	CKR06							
C108	17075	CLR79T2							
C109	3300/50	CKR05							
C112	22/200	HV22							
C113	.1/100	CKR06							
C114	22/200	HV21							
C115	.01/200	CKR06							
C116	22/200	HV22							
C117	22/200	HV22							
C201	.082/500	HV20							
C202	.082/500	HV20							
C203	.082/500	HV20							
C204	.082/500	HV20							
C205	.047/1KV	HV21							
C206	.047/1KV	HV21							
C207	.047/1KV	HV21							
C208	.047/1KV	HV21							
C209	.015/2KV	HV23							
C210	.015/2KV	HV23							
C211	.015/2KV	HV23							
C212	.015/2KV	HV23							
C213	.015/2KV	HV23							
C214	.015/2KV	HV23							
C215	.015/2KV	HV23							
C216	4700P/2KV	HV21							
C217	4700P/2KV	HV21							
C218	4700P/2KV	HV21							
C219	4700P/2KV	HV21							
C220	4700P/2KV	HV21							
C221	4700P/2KV	HV21							
C222	4700P/2KV	HV21							
C223	4700P/2KV	HV21							
C224	4700P/2KV	HV21							
C225	4700P/2KV	HV21							
C226	4700P/2KV	HV21							
C227	4700P/2KV	HV21							
C228	4700P/7.5KV	HS33							

Ref	Value	Comment	Part Number	Mfg	Ven	PO	Cost	Date Code	S/N
C229	4700P/7.5KV	HS33							
C230	10/50V	CLR-79T2							
C231	3300/100	CKR05							
C232	120/500	HV21							
C233	10/50V	CLR-79T2							
C234	3300/100	CKR05							
C235	330/500	HV21							
C236	.1/100	CKR06							
C237	.1/100	CKR06							
C401	22/35	CSR-21							
C402	22/35	CSR-21							
C403	15/30	CLR79T1							
C404	.1/100	CKR06							
C405	.1/100	CKR06							
C406	.1/100	CKR06							
C407	.1/100	CKR06							
C408	.1/100	CKR06							
C409	.15/800	HV23							
C410	.01/200	CKR06							
C411	33/15	CLR79T1							
C412	33/15	CLR79T1							
C413	.1/100	CKR06							
C414	.1/100	CKR06							
D101	1N5617	BODY-A							
D102	1N5617	BODY-A							
D103	1N751-1	DO-35							
D201	F30	F30							
D202	F30	F30							
D203	F30	F30							
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D211	F30	F30							
D212	F30	F30							
D213	F30	F30							
D214	SAT	F30							
D215	SAT	F30							
D216	F30	F30							
D217	F30	F30							

Ref	Value	Comment	Part Number	Mfg	Ven	PO	Cost	Date Code	S/N
D218	F30	F30							
D219	F30	F30							
D220	F30	F30							
D221	F30	F30							
D222	F30	F30							
D223	F30	F30							
D224	F30	F30							
D225	F30	F30							
D226	F30	F30							
D227	F30	F30							
D228	F30	F30							
D229	1N6318	DO-35							
D230	1N4148-1	DO-35							
D231	1N4148-1	DO-35							
D232	1N4148-1	DO-35							
D233	F30	F30							
D401	1N4148-1	DO-35							
D402	1N4148-1	DO-35							
D403	1N4148-1	DO-35							
D404	1N4148-1	DO-35							
D405	1N4148-1	DO-35							
D406	1N4148-1	DO-35							
D407	F30	F30							
D408	1N4995	BODY-B							
D409	1N4148-1	DO-35							
FB101	2673000101	Ferrite Bead							
FB201	2673000101	Ferrite Bead							
FB202	2673000101	Ferrite Bead							
H101	AE10198								
J7A		OMNETICS							
J7B		OMNETICS							
MA1	061-0349	Mag. Assembly							
Q101	2N3700	TO-18							
Q102	SFT1192-5	TO-5							
Q103	2N3700	TO-18							
Q104	2N2907	TO-18							
Q201	2N2907	TO-18							
Q202	2N2222	TO-18							
Q203	2N3700	TO-18							
Q204	SFT1192-5	TO-5							
Q205	2N2907	TO-18							
Q206	2N2222	TO-18							
Q207	2N3700	TO-18							

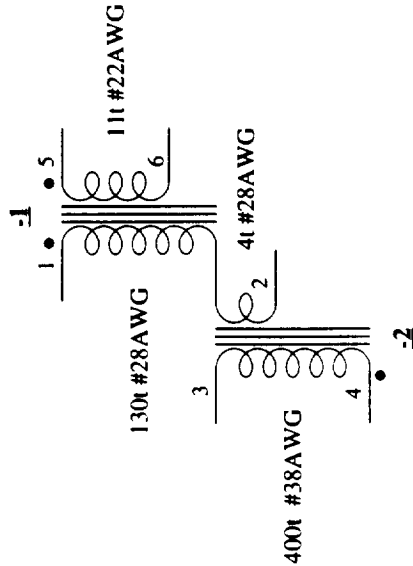
Ref	Value	Comment	Part Number	Mfg	Ven	PO	Cost	Date Code	S/N
Q401	SFT1192-5	TO-5							
Q402	SFT1192-5	TO-5							
Q403	2N2907	TO-18							
Q404	2N3700	TO-18							
Q405	2N3700	TO-18							
Q406	2N3700	TO-18							
Q407	2N2907	TO-18							
R101	13.3K*	RNC50							
R102	13.3K*	RNC50							
R103	22K	RNC50							
R104	100K	RNC50							
R105	120.0*	RNC50							
R106	4.7K	RCR05							
R106A	47K	RCR05							
R107	470	RCR05							
R108	470	RCR05							
R109	110K	RNC50							
R110	SAT	RNC50							
R111	47	RCR05							
R112	470	RCR05							
R113	470	RCR05							
R114	470	RCR05							
R115	SAT	RNC50							
R116	SAT	RNC50							
R117	SAT	RNC50							
R118	SAT	RNC50							
R119	SAT	RNC50							
R120	SAT	RNC50							
R122	2M	RCR07							
R123	5M	MG660							
R124	SAT	RNC50							
R125	SAT	RNC50							
R126	SAT	RNC50							
R127	SAT	RNC50							
R128	SAT	RNC50							
R129	SAT	RNC50							
R130	68.00	RCR05							
R201	22K	RNC50							
R202	110K	RNC50							
R203	SAT	RNC50							
R204	10K	RCR05							
R205	47	RCR07							
R206	680	RCR05							

Ref	Value	Comment	Part Number	Mfg	Ven	PO	Cost	Date Code	S/N
R207	10K	RCR07							
R208	50M	MOX400							
R209	10K	RCR07							
R210	4.7K	RCR05							
R211	15K	RCR05							
R212	22K	RNC50							
R213	110	RNC50							
R214	SAT	RNC50							
R215	100K	RNC50							
R216	SAT	RNC50							
R217	10K	RCR05							
R218	47	RCR07							
R219	680	RCR05							
R220	100K	RCR07							
R221	100K	RCR07							
R222	1K	RCR07							
R223	1K	RCR07							
R224	SAT	RNC50							
R225	SAT	RNC50							
R226	SAT	RNC50							
R227	SAT	RNC50							
R228	110K	RNC50							
R301	18.7K	RNC50							
R302	200K	RNC50							
R303	SAT	RNC50							
R304	499K	RNC50							
R305	499K	RNC50							
R306	18.7K	RNC50							
R307	SAT	RNC50							
R308	18.7K	RNC50							
R309	30K	YSI44008							
R405	330	RCR05							
R406	47K	RCR05							
R407	47K	RCR05							
R408	1K	RCR05							
R409	1M	RCR05							
R410	TBD	RCR07							
R411	100K	RCR07							
R412	27K	RNC50							
R413	10K	RCR05							
R414	47K	RCR05							
R415	10M	MG660							
R416	226K	RNC55							

Ref	Value	Comment	Part Number	Mfg	Ven	PO	Cost	Date Code	S/N
R417	90.9K	RNC50							
R418	93.1K	RNC50							
R419	162K	RNC50							
R420	49.9K	RNC50							
R422	499K	RCR07							
R423	49.9K	RNC50							
R424	976K	RNC50							
T101	061-0351	1408 Core							
T201	061-0350	1408 Core							
T202	061-0364	1408 Core							
U401	1/2RH1013M	8-DIP							
U402	UC1709	8-DIP							
U403	1/2RH1013M	8-DIP							
X101	2031B Usec0	Turret							
X102	2031B Usec0	Turret							
X103	2031B Usec0	Turret							
X104	2031B Usec0	Turret							
X105	2031B Usec0	Turret							
X106	2031B Usec0	Turret							
X107	2031B Usec0	Turret							
X108	2031B Usec0	Turret							
X109	2031B Usec0	Turret							
X110	2031B Usec0	Turret							
X111	2031B Usec0	Turret							
X112	2031B Usec0	Turret							
X113	2031B Usec0	Turret							
X114	2031B Usec0	Turret							
X115	2031B Usec0	Turret							
X116	2031B Usec0	Turret							
X117	2031B Usec0	Turret							
X118	2031B Usec0	Turret							
X119	2031B Usec0	Turret							
X120	2031B Usec0	Turret							
X121	2031B Usec0	Turret							
X122	2031B Usec0	Turret							
X123	2061B Usec0	Turret							
X124	2061B Usec0	Turret							
X125	2031B Usec0	Turret							
X126	2031B Usec0	Turret							
X130	2031B Usec0	Turret							
X131	2031B Usec0	Turret							
X132	2061B Usec0	Turret							
X133	2061B Usec0	Turret							

Ref	Value	Comment	Part Number	Mfg	Ven	PO	Cost	Date Code	S/N
X134	2061B Useco	Turret							
X135	2061B Useco	Turret							
X136	2061B Useco	Turret							
X137	2061B Useco	Turret							
X138	2061B Useco	Turret							
X139	2061B Useco	Turret							
X140	2061B Useco	Turret							
X141	2061B Useco	Turret							
X142	2061B Useco	Turret							
X143	2061B Useco	Turret							
X144	2061B Useco	Turret							
X145	2031B Useco	Turret							
X146	2031B Useco	Turret							
X147	2031B Useco	Turret							
X148	2031B Useco	Turret							
X149	2031B Useco	Turret							
X150	2031B Useco	Turret							
X151	2031B Useco	Turret							
X152	2031B Useco	Turret							
X153	2031B Useco	Turret							
X154	2031B Useco	Turret							
X155	2031B Useco	Turret							
X156	2031B Useco	Turret							
X157	2031B Useco	Turret							
X201	2031B Useco	Turret							
X202	2031B Useco	Turret							
X301	2031B Useco	Turret							
X302	2031B Useco	Turret							
X303	2031B Useco	Turret							
X304	2031B Useco	Turret							
X401	2031B Useco	Turret							
X402	2031B Useco	Turret							
X403	2031B Useco	Turret							
X404	2031B Useco	Turret							
X405	2031B Useco	Turret							
X406	2031B Useco	Turret							
X407	2031B Useco	Turret							
X408	2031B Useco	Turret							
X409	2031B Useco	Turret							
X410	2031B Useco	Turret							

WINDING INSTRUCTIONS



LIST OF MATERIALS

Item	QTY	P/N	Description
1	1	768T188-3F3	Philips/Ferroxcube Toroid
2	1	TBD	Robert Hadley Tape Wound Core
3	AR	HAPT-22	#22AWG Magnet Wire
4	AR	HAPT-28	#28AWG Magnet Wire
5	AR	EPON 815/Z	Encapsulate
6	AR	EC2850FT/CAT 9	Epoxy (Black)

Notes:

- 1) Core -1 is Philips/Ferroxcube 768T188-3F3.
Core -2 is Magnetics 80550-1/4D, prewound by Robert Hadley.
- 2) All windings in same direction around core.

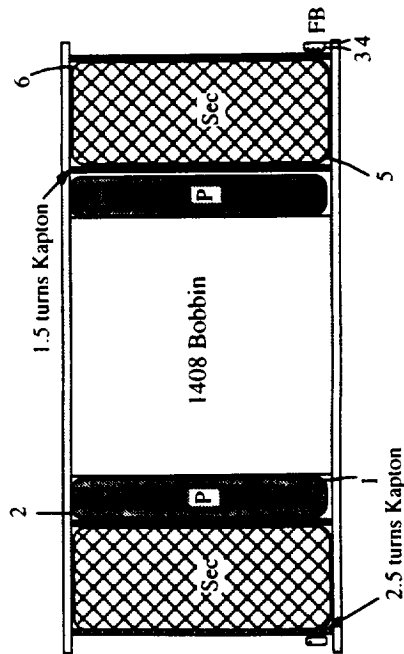
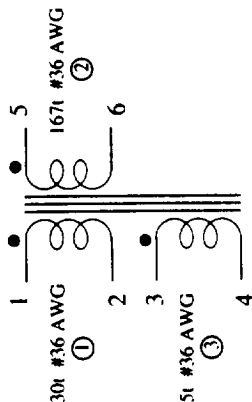
Space Physics Research Laboratory College of Engineering University of Michigan		Filament/Current Transformers Filament Bias Supply Huygens - GCMS	
061-NMS	0TK63	J0349-Fil/Bias AC Trans	061-0349
DATE	BY	CHECKED	APPROVED
31 JUL 95	O. Beryl		
RELEASE	DATE	BY	REASON
31 JUL 95	O. Beryl		

LIST OF MATERIALS

Item	QTY	P/N	Description
1	1	1408PA250-3B7	Philips Magnetic Core
2	1	1408F1D	Philips Plastic Bobbin
3	AR	EPON 815/V140	Encapsulate

NOTES:

- 1) ○ denotes winding order.
- 2) ● denotes phasing.
- 3) 1.5 layers of Kapton tape between 1 and 2.
2.5 layers of Kapton tape between 2 and 3.

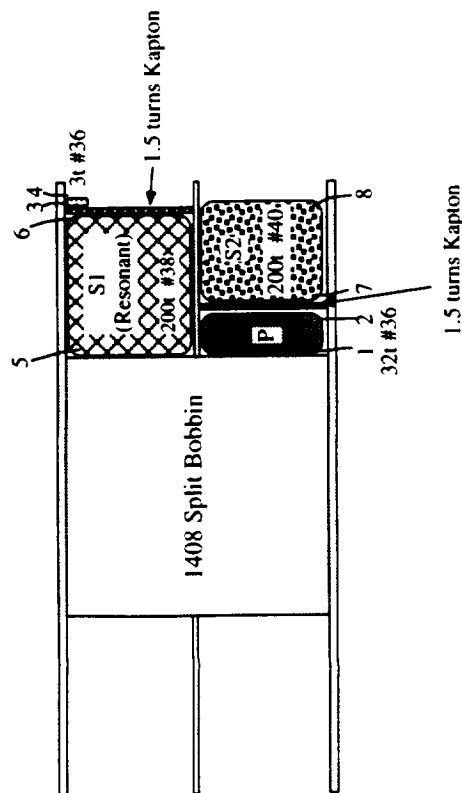
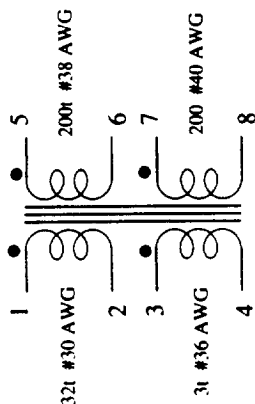


Space Physics Research Laboratory College of Engineering University of Michigan		FB -780 Bias Transformer Filament Bias Supply Planet - B NMS	
061-NMS	0TK63	J0350-FB -780 Bias Xfm.	061-0350
DRAWN: O. Bavi		CHECKED: S. Baret/K. Almet	
RELEASE: 1-Sep-94		DATE: 1-Sep-94	
PRINTED: 2-7-1996 02:50 PM		DRAWING PAGE: 5.21	

LIST OF MATERIALS

Item	QTY	P/N	Description
1	1	1408PA250-3B7	Philips Magnetic Core
2	1	1408F1D	Philips Plastic Bobbin
3	AR	EPON 815/V140	Encapsulate

NOTES:



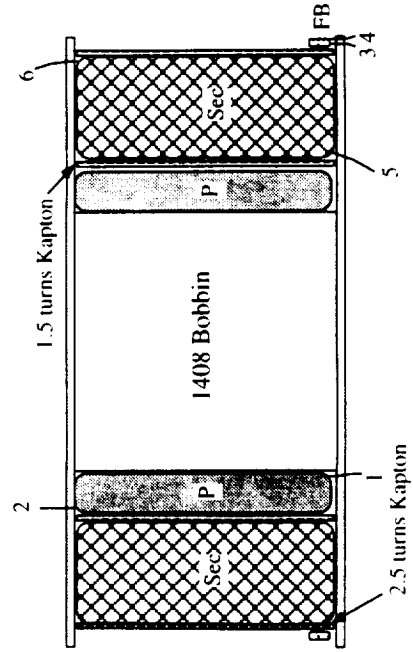
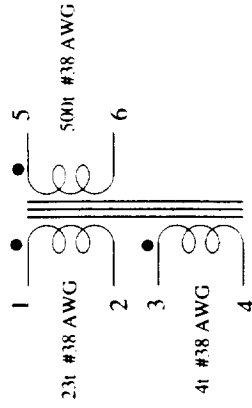
Space Physics Research Laboratory
College of Engineering
University of Michigan

FB +/-75V Bias Transformer
Filament Bias Supply
Planet B - NMS

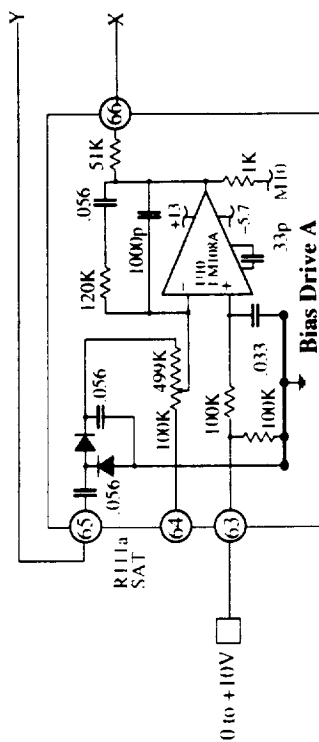
061-NMS	OTK63	J0351-FB +/-75 Bias Xfrm.	061-0351
DATE		2/27/96 03:00 PM	Printed: 2
DRAWN		S. Burt/K. Arnold	Drawing Page
CHECKED		S. Burt/K. Arnold	5.22
RELEASE		24 Aug 93	
INSTR		O. Burt	

LIST OF MATERIALS

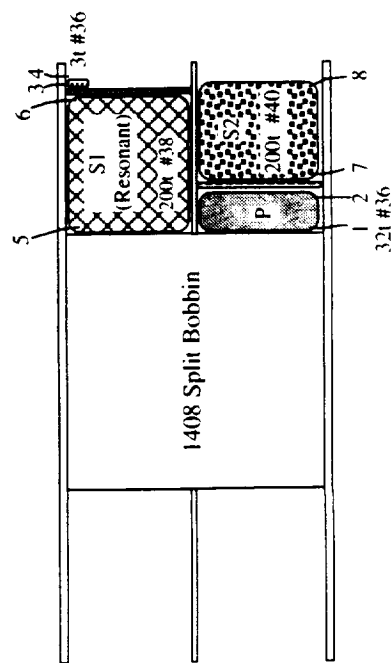
Item	QTY	P/N	Description
1	1	1408PA160-3B7	Philips Magnetic Core
2	1	1408F1D	Philips Plastic Bobbin
3	AR	EPON 815/V140	Encapsulate



Space Physics Research Laboratory College of Engineering University of Michigan		FB EM Transformer Planet - B NMS		061-NMS		0TK63 J0364-FB EM Transformer		061-0364		RELEASE 30-Sep-95 DRAWN: 00083 J May-95 CHECKED: 00083 J May-95 DESIGNED: 00083 J May-95 DRAWING: 00083 J May-95	
Printed		2		12/28/96		12/28/96		1		Drawing Page	
5.23											



"A" Transformer Configuration



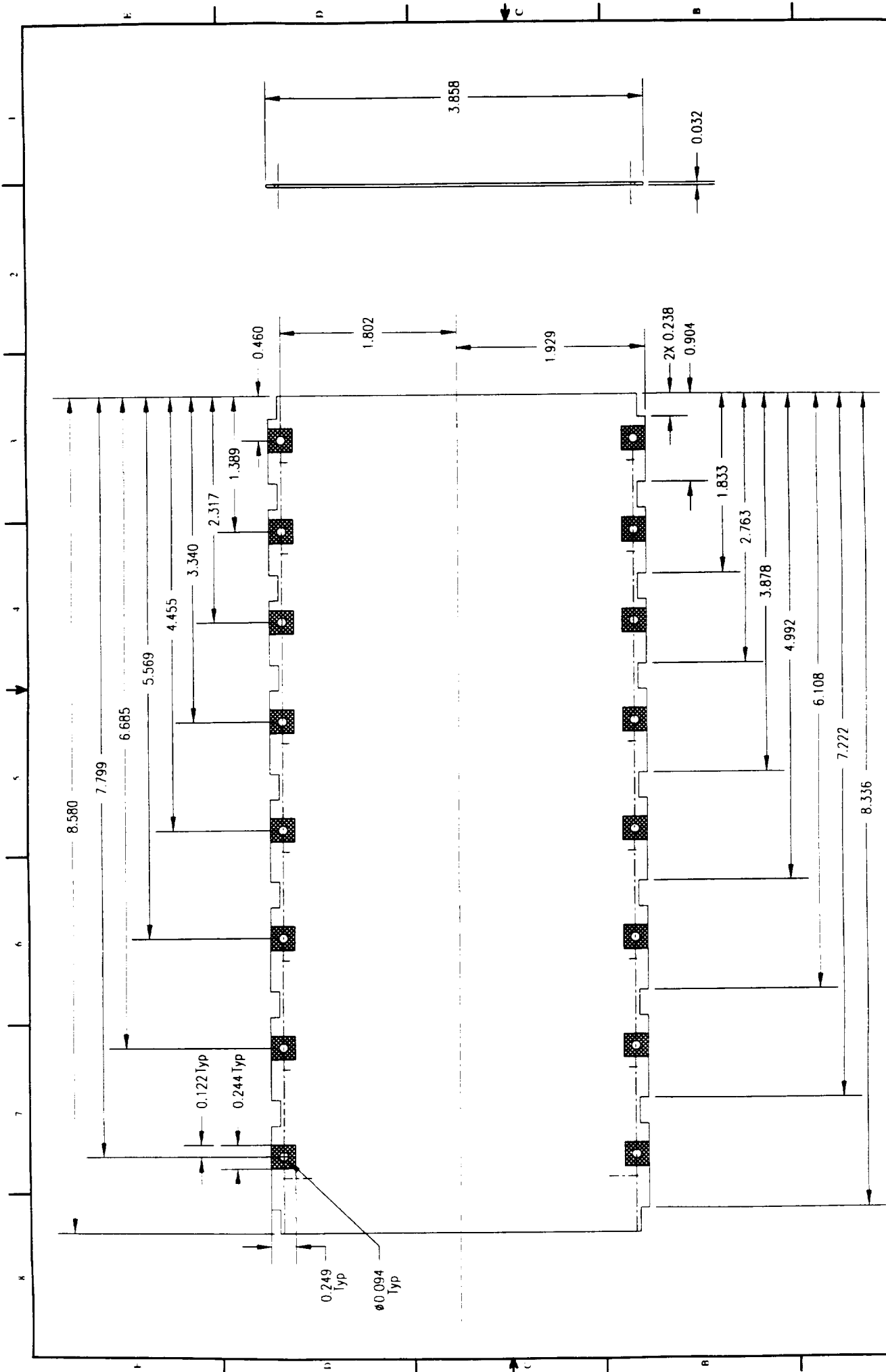
Space Physics Research Laboratory
College of Engineering
University of Michigan

**FB $\pm 75V$ Bias
Test Oscillator Circuit
Planet - B NMS**

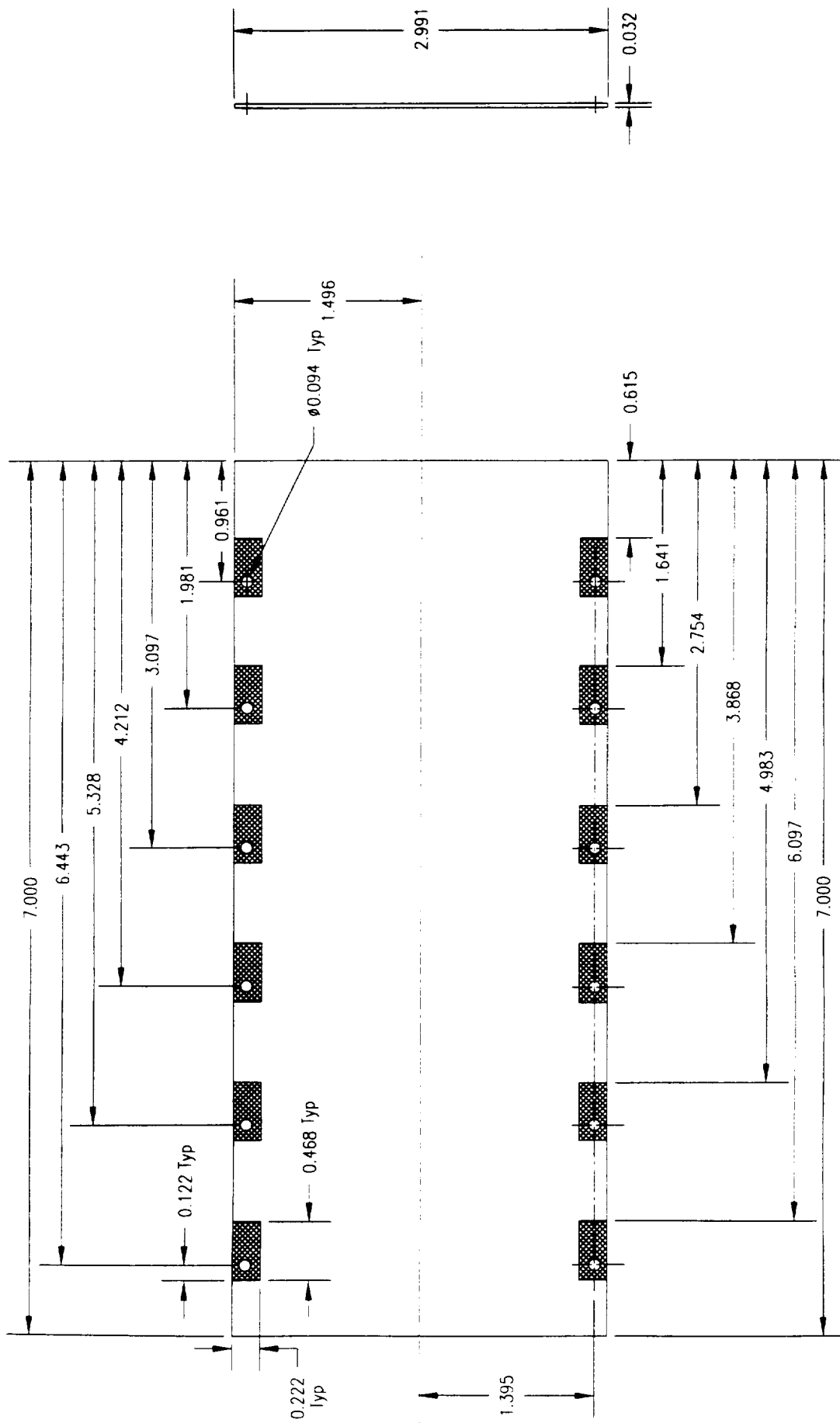
UNIT NAME	UNIT CODE	UNIT TITLE	UNIT CREDIT
061-NMS	0TK63	J0356A-Bias Test Osc A	061-0356

Printed 2
MILITARY 1
Drawing Page 1

5.40



Space Physics Research Laboratory College of Engineering University of Michigan		Outer PCB Outline FB Planet - B NMS		Drawing Page 5.49	
Planet B - 080249 OTK63 J0396A-FB Outer PCB		061-0396 A		Released: 29 Feb 96 Drawn: 29 Feb 96 from ACAD (COT/ENG) Baitel / Aneet	
Printed: 2 02/29/96 08:28					



Space Physics Research Laboratory College of Engineering University of Michigan		Inner PCB Outline FB Planet - B NMS	
Planet B - 080249 OTK63 J0397A-FB Inner PCB Outline		Planet B - 080249 OTK63 J0397A-FB Inner PCB Outline	
061-0397		061-0397	
A		A	
DATE: 01/25/96		DATE: 01/25/96	
BY: J. J. J.		BY: J. J. J.	
CHECKED: J. J. J.		CHECKED: J. J. J.	
RELEASE: 29-Feb-96		RELEASE: 29-Feb-96	
DRAWN: 29-Feb-96 (from ACAD)		DRAWN: 29-Feb-96 (from ACAD)	
CUSTODY: Planet/Annet		CUSTODY: Planet/Annet	

Section 6

RF - RF Module

**Linear
Regulator
Reference**

RF +15 [1] RF 15 Gnd [6] +8.4 Ref

-RF AGC [1]

RF Temp [1]

RF -15 [1] 30K YSI

RF +130 [4] -15

RF -130 [4] +130

RF +5 [2] -130

RF 5 Gnd [7] +5

**Serial
to
Parallel**

RF Data [14]

RF Clk [15]

RF AC Enb [2]

RF DC Enb [7]

POR [12]

RF Cal [4]

RF Logic Gnd [6]

J5A J5B

**10 bit
D/A**

Ref

+5

**1 bit
Latch**

+5

**10 bit
D/A**

Ref

+5

RF Amplitude Feedback

2 Frequency Resonant
Tank Circuit
(1.9 MHz & 3.9 MHz)

A
Amplitude
Control

A
Amplitude
Control

**Switched
Band-Pass
Filter**

Frequency Select

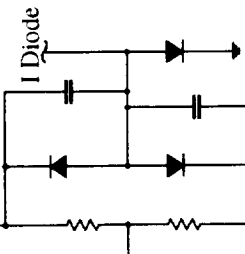
**HV Op-
Amp**
- Rod DC
(~17%)

**HV Op-
Amp**
+ Rod DC

Frequency
Count

- Rod

+ Rod
($V_{RMS} \approx 600V$ max)



Space Physics Research Laboratory
College of Engineering
University of Michigan

RF Functional Block Diagram
Planet - B NMS

061-NMS 0TK63 J0141-RF Block

061-0141

COOLING BLOCK

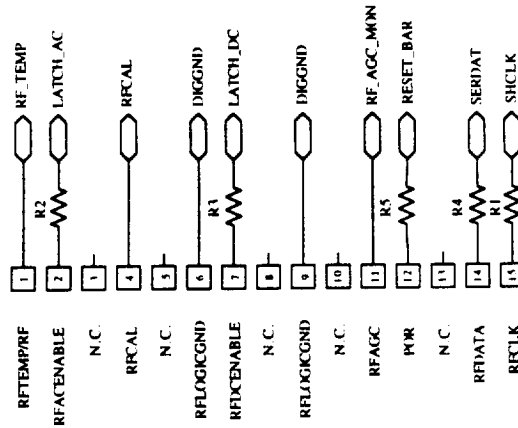
RELEASE 12-MAY-95
DRAWN 10-MAY-95
AUG 95

6.02

Printed 4:39:42 PM

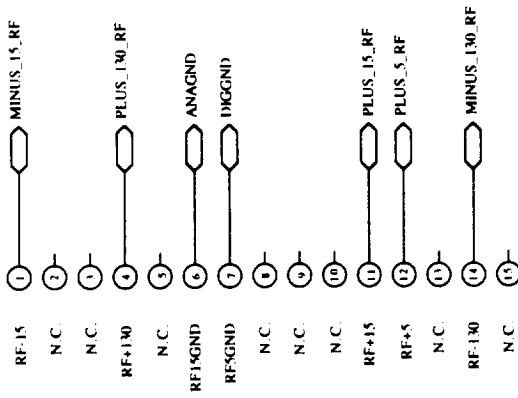
Drawing Page

J1
HD22-15S

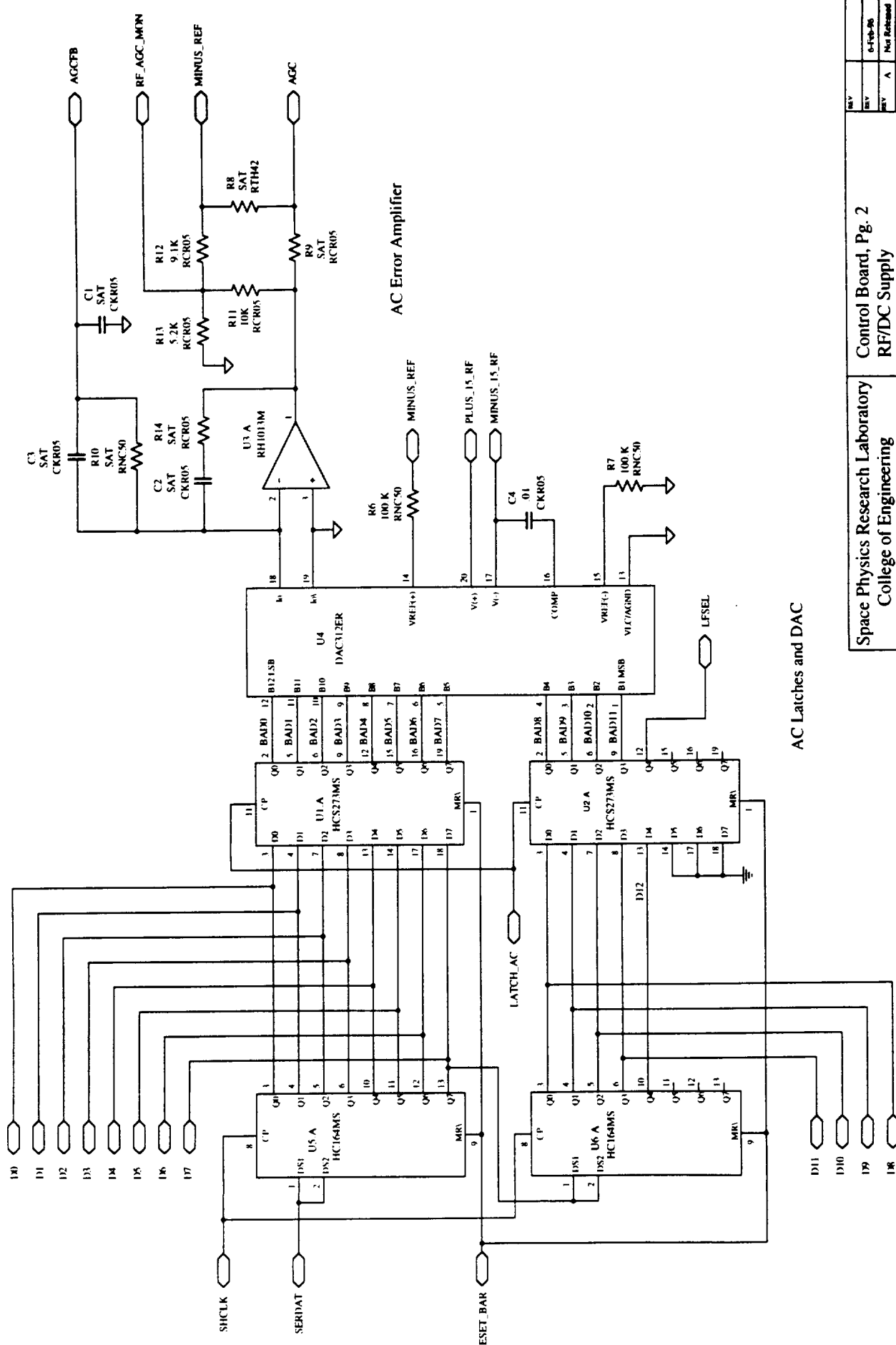


Note: R1 through R5 are 1.2K RCRO5.

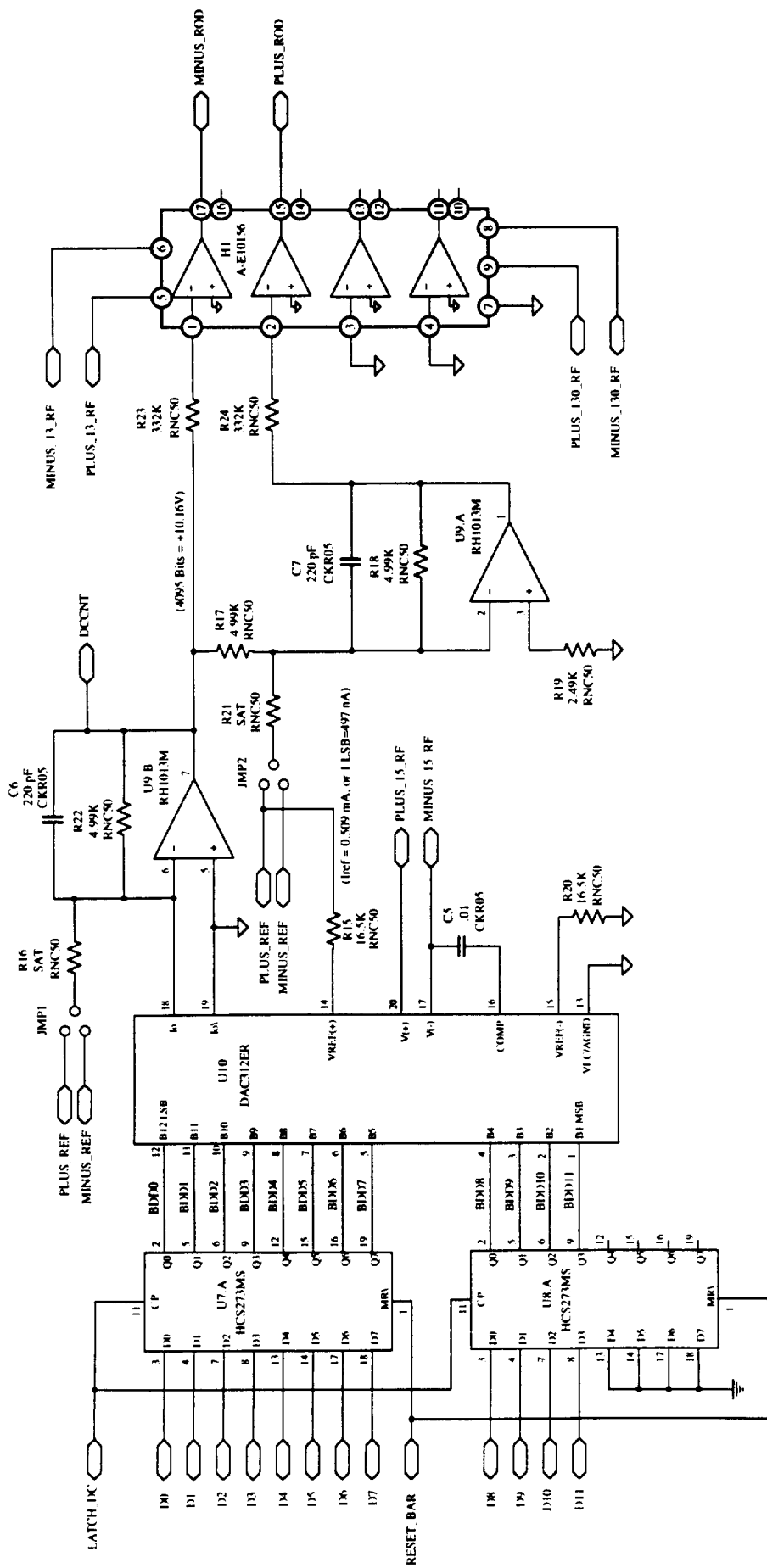
J2
HD22-15P



Space Physics Research Laboratory College of Engineering University of Michigan		Control Board, Pg. 1 RF/DC Supply Planet-B NMS		REV	REV	REV	REV	REV
Planet B - NMS		J0355.1A-RF/DC Ctrl. Pg. 1		061-0355	A	061-0355	A	061-0355
J0355.1A-RF/DC Ctrl. Pg. 1		J0355.1A-RF/DC Ctrl. Pg. 1		061-0355	A	061-0355	A	061-0355
J0355.1A-RF/DC Ctrl. Pg. 1		J0355.1A-RF/DC Ctrl. Pg. 1		061-0355	A	061-0355	A	061-0355
J0355.1A-RF/DC Ctrl. Pg. 1		J0355.1A-RF/DC Ctrl. Pg. 1		061-0355	A	061-0355	A	061-0355
J0355.1A-RF/DC Ctrl. Pg. 1		J0355.1A-RF/DC Ctrl. Pg. 1		061-0355	A	061-0355	A	061-0355
J0355.1A-RF/DC Ctrl. Pg. 1		J0355.1A-RF/DC Ctrl. Pg. 1		061-0355	A	061-0355	A	061-0355
J0355.1A-RF/DC Ctrl. Pg. 1		J0355.1A-RF/DC Ctrl. Pg. 1		061-0355	A	061-0355	A	061-0355
J0355.1A-RF/DC Ctrl. Pg. 1		J0355.1A-RF/DC Ctrl. Pg. 1		061-0355	A	061-0355	A	061-0355
J0355.1A-RF/DC Ctrl. Pg. 1		J0355.1A-RF/DC Ctrl. Pg. 1		061-0355	A	061-0355	A	061-0355



Space Physics Research Laboratory College of Engineering University of Michigan		Control Board, Pg. 2 RF/DC Supply Planet-B NMS		REV	6-Feb-86
Planet B - NMS		J0355.2A-RF/DC Ctrl, Pg. 2		REV	A
07TK63		061-0355		RELEASE	20-Sep-95
PROJECT		DRAWN		DATE	24-Aug-95
061-0355		A		BY	B. BLACK

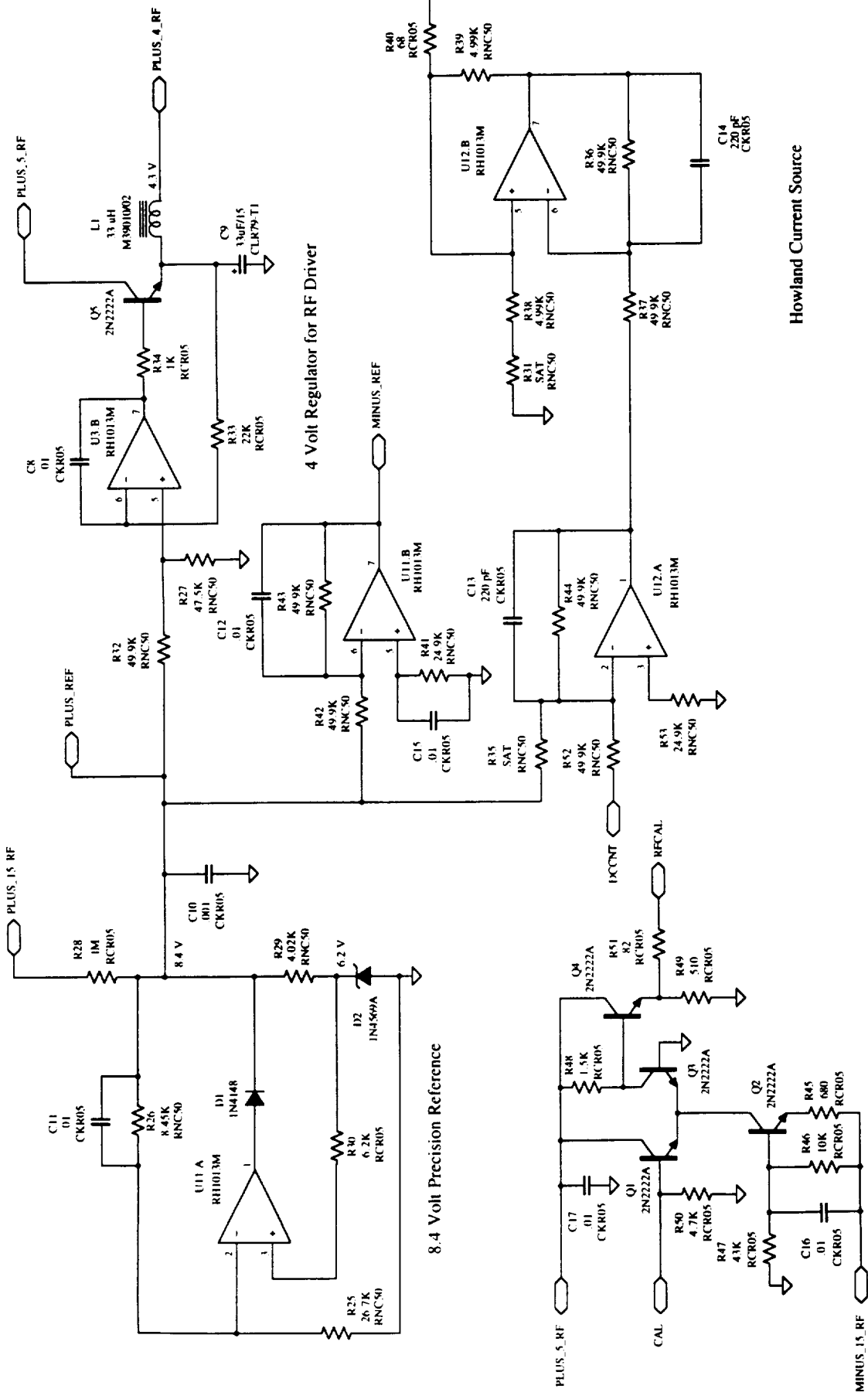


DC Latches and DAC

DC Control Voltage Amplifiers

High Voltage Amplifiers

Space Physics Research Laboratory		Control Board, Pg. 3	
College of Engineering		RF/DC Supply	
University of Michigan		Planet-B NMS	
Planet B - NMS	07K63	J0355-2A-RF/DC Ctrl, Pg. 3	061-0355
REV	REV	REV	REV
0	0	0	0
1	1	1	1
2	2	2	2
3	3	3	3
4	4	4	4
5	5	5	5
6	6	6	6
7	7	7	7
8	8	8	8
9	9	9	9
10	10	10	10
11	11	11	11
12	12	12	12
13	13	13	13
14	14	14	14
15	15	15	15
16	16	16	16
17	17	17	17
18	18	18	18
19	19	19	19
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21	21	21	21
22	22	22	22
23	23	23	23
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92	92	92	92
93	93	93	93
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99	99	99	99
100	100	100	100



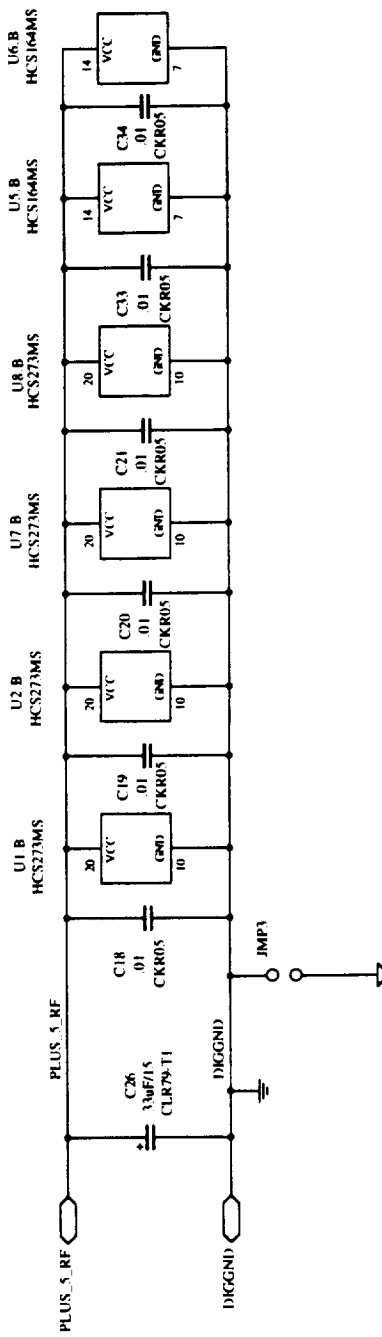
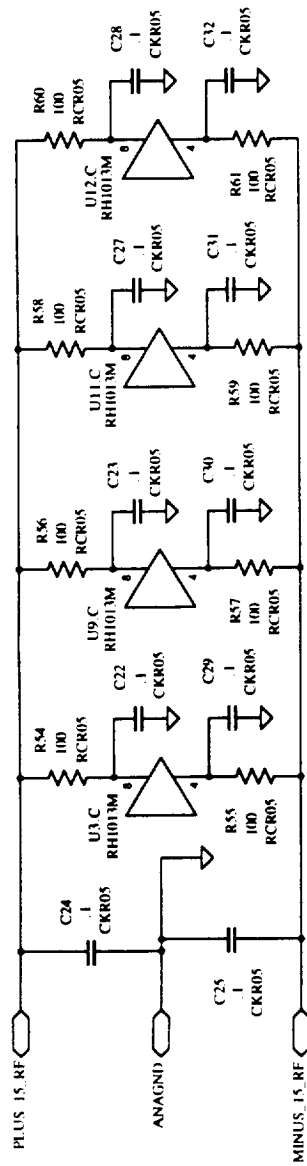
RF Calibration Driver

Howland Current Source

4 Volt Regulator for RF Driver

8.4 Volt Precision Reference

Space Physics Research Laboratory		Control Board, Pg. 4	
College of Engineering		RF/DC Supply	
University of Michigan		Planet-B NMS	
PROJECT	Planet B - NMS	REVISION	0061-0355
DATE	07K63	DESIGNED BY	A
APPROVED BY		DATE	
6-Feb-96		6-Feb-96	
Not Released		Not Released	
RELEASE		RELEASE	
20-Sep-95		20-Sep-95	
INAWN		INAWN	
24-Aug-95		24-Aug-95	
CYCLIC		CYCLIC	
B. BLOCK		B. BLOCK	



Space Physics Research Laboratory College of Engineering University of Michigan	Control Board, Pg. 5 RF/DC Supply Planet-B NMS
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Control Board, Pg. 5
RF/DC Supply
Planet-B NMS

PROJECT	NAME	PHONE	TELEPHONE NUMBER
Planet B - NMS	0TK63	J0355.5A-RF/DC Ctrl, Pg. 5	061-0355

0355

61-0

5	0
---	---

1, Pg.

C Ctrl

RF/D

5A-1

10355

63	J
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ΣΤΙΚ
ΕΛΛΗΝΙΚΗ ΔΗΜΟΚΡΑΤΙΑ

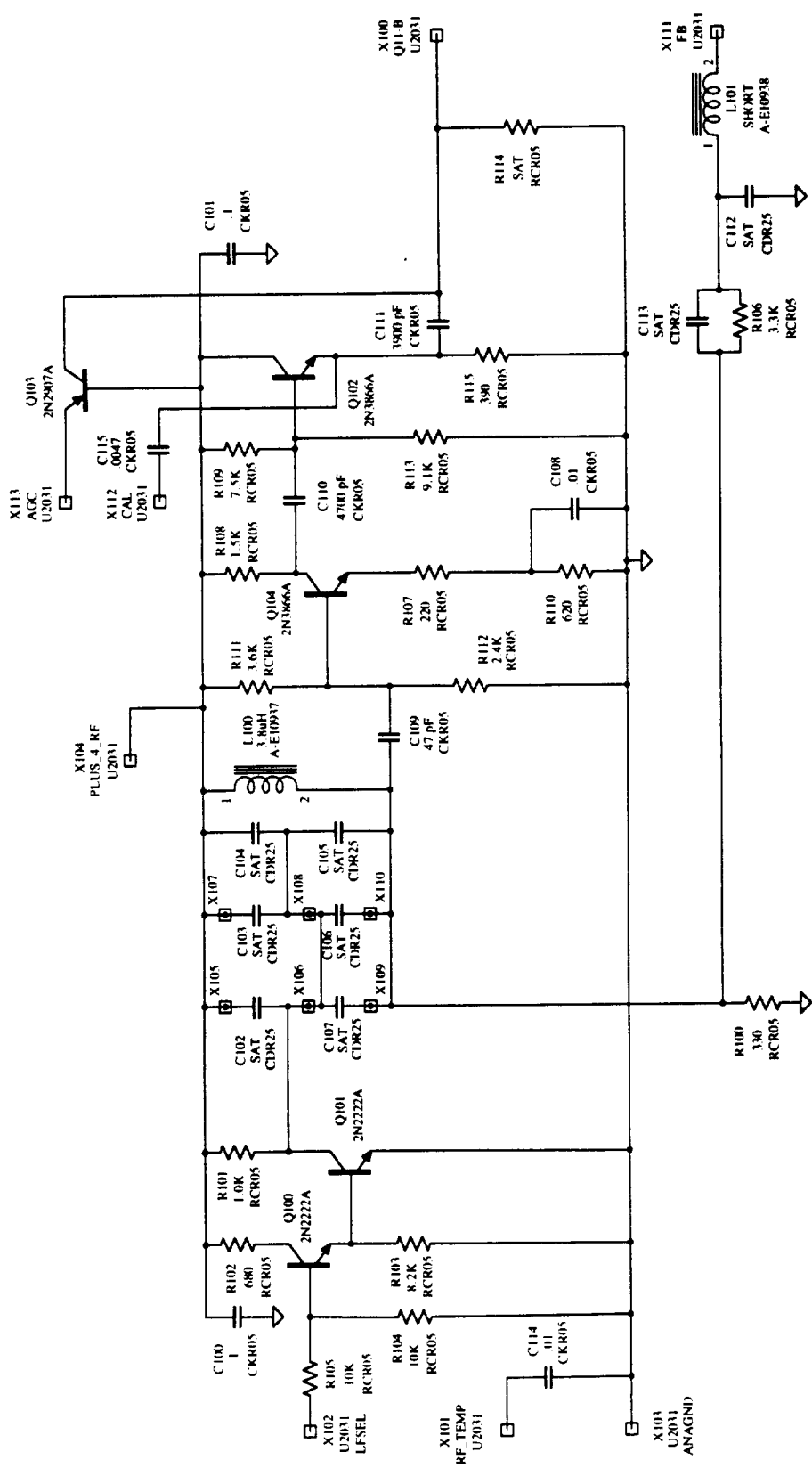
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NM

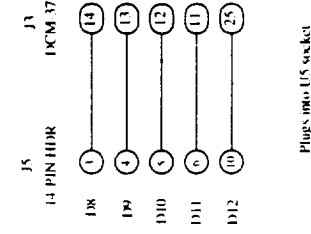
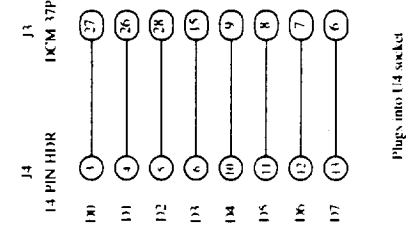
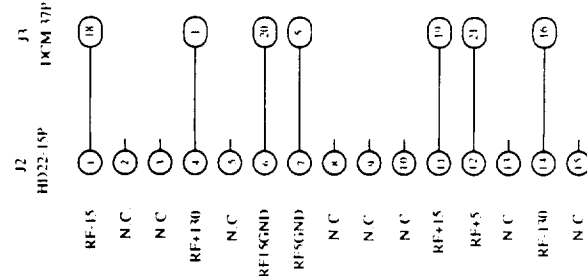
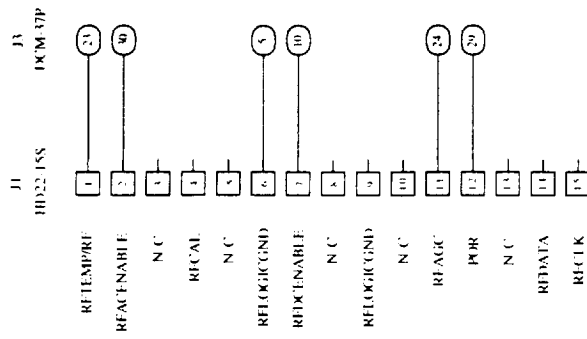
et B -

Plan

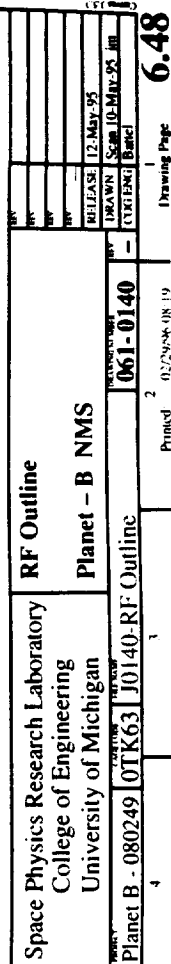
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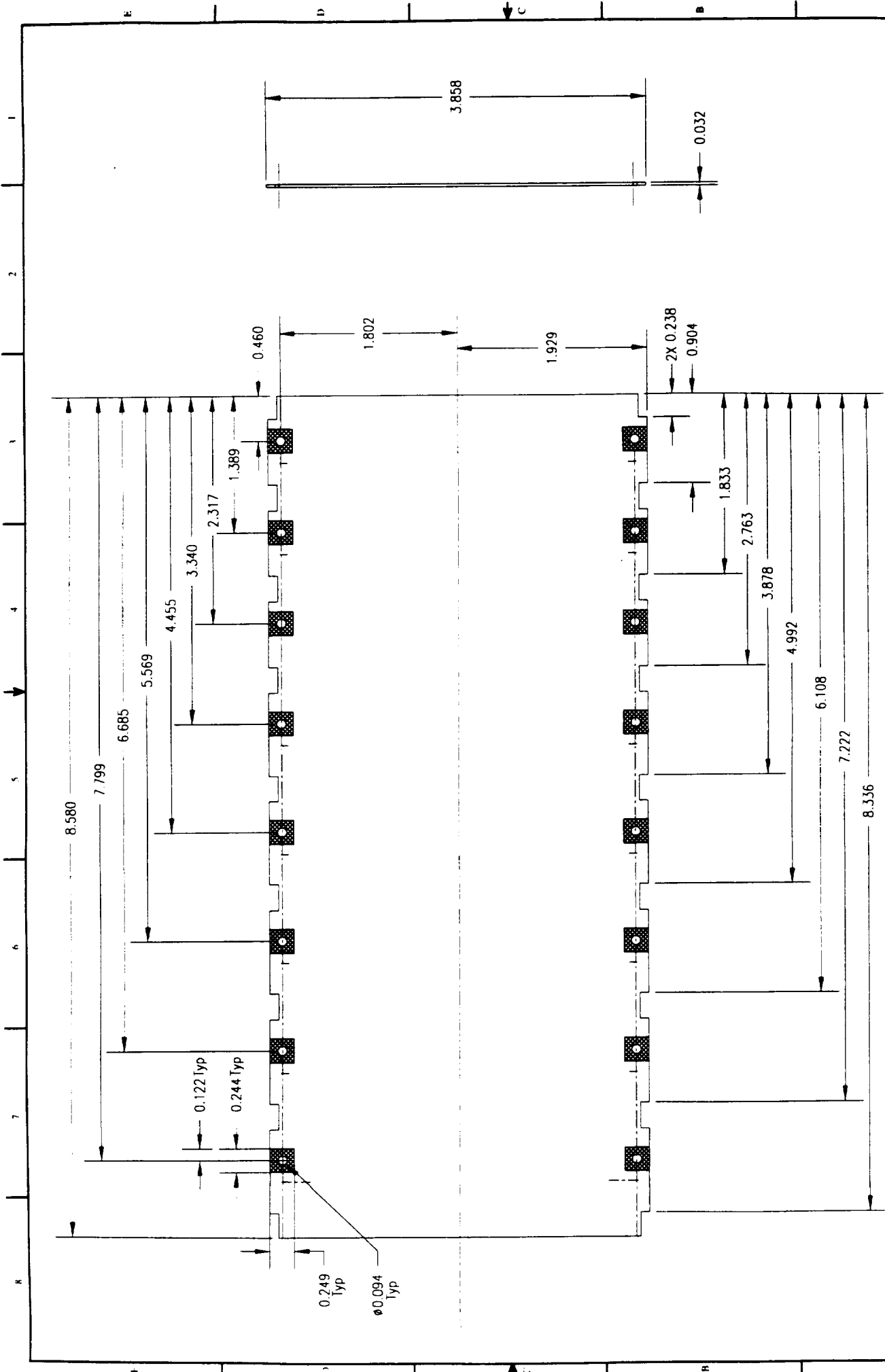


Space Physics Research Laboratory		RF/DC Driver Board	
College of Engineering		RF/DC Supply	
University of Michigan		Planet-B NMS	
Planet B - NMS	0TK63 J0369A	RF/DC RF Driver Bd	061-0369
REV	REV	REV	REV
7-Sep-95	7-Sep-95	7-Sep-95	7-Sep-95
Not Released	Not Released	Not Released	Not Released
RELEASE	RELEASE	RELEASE	RELEASE
12-Oct-95	12-Oct-95	12-Oct-95	12-Oct-95
DRAWN	DRAWN	DRAWN	DRAWN
9/11/95 DB	9/11/95 DB	9/11/95 DB	9/11/95 DB
BY	BY	BY	BY
ENG	ENG	ENG	ENG

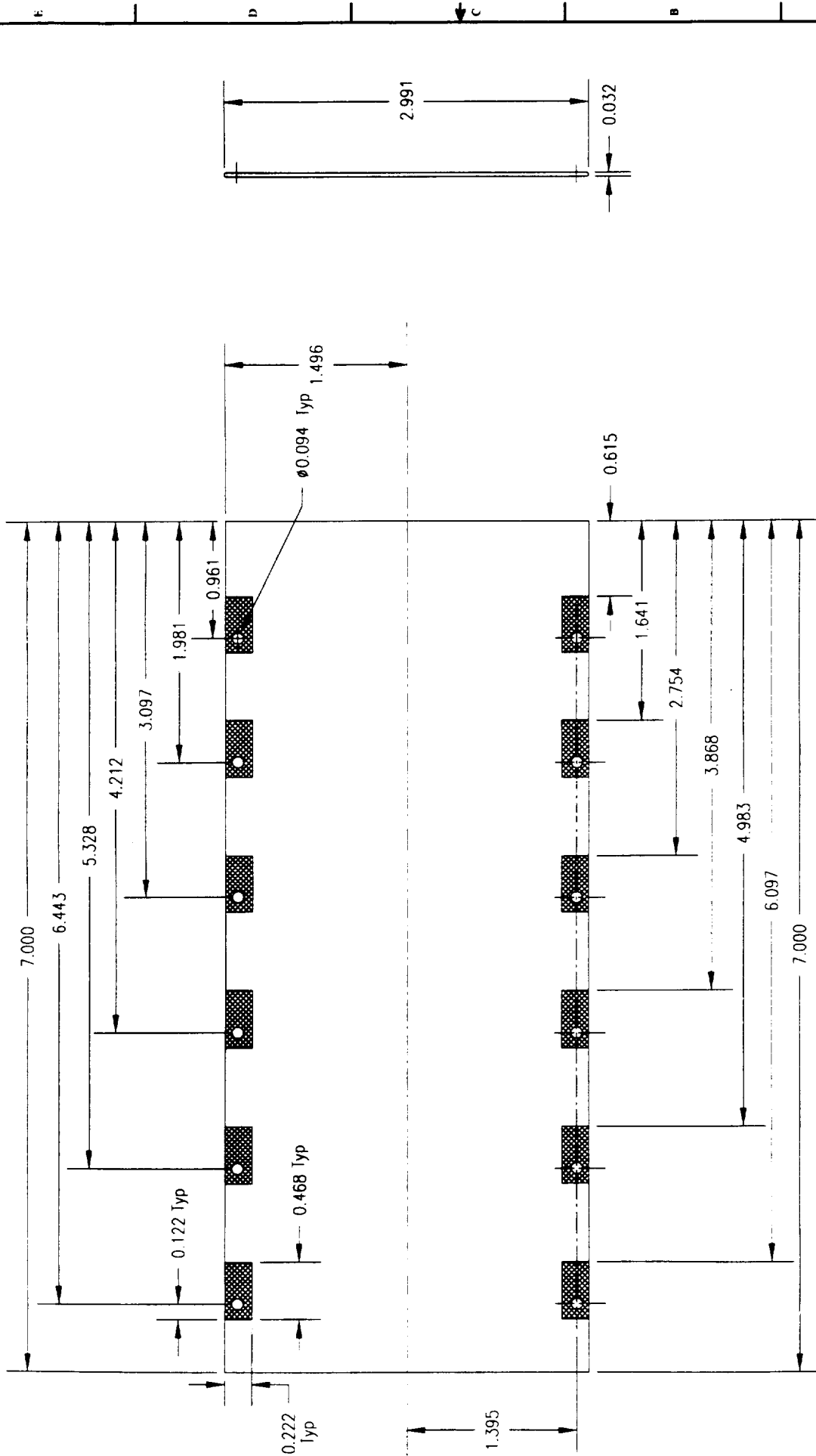


Space Physics Research Laboratory College of Engineering University of Michigan		Test Box Adapter Harness RF/DC Supply Planet-B NMS		REV	
				REV	
				REV	
				RELEASE	12 Oct 95
				DRAWING	12 Sept 95
				CHECKING	B BLACK
Planet B - NMS		J0373	RF/DC BB Harness	061-0373	
01K63	REVISIONS				





Space Physics Research Laboratory College of Engineering University of Michigan		Outer PCB Outline RF Planet - B NMS		Drawing Page 1	
Planet B - 080249		J0398A-RF Outline		061-0398 A	
DATE: 02/24/86		DESIGNED BY: J0398A		CHECKED BY: J0398A	
DRAWN: J0398A		RELEASE: 20-Feb-96		20-Feb-96 from ACAD	
TUG: BKT		BKT / Amsel		6.49	



Space Physics Research Laboratory		Inner PCB Outline	
College of Engineering		RF	
University of Michigan		Planet - B NMS	
Planet B - 080249	OTK63	J0399A-RF: Outline	061-0399
DATE: 29-SEP-96		DRAWN: 29-SEP-96 (J0399A)	
CHECKED: 29-SEP-96 (J0399A)		APPROVED: 29-SEP-96 (J0399A)	
PRINTED: 02/29/96 (J0399A)		DRAWING PAGE: 6.50	

Section 7

CM - Control Module

J2
PIM

+5
On/Off pulses for "power switch"

CLK

Command

Address/Cmnnds

Control

Data

Address

Control

Data

Interface

Memory

PROM

Sequential
Controller
(FPGA)

8-bit
ADC

8-bit
DAC

Data

Ctrl

Threshold Detectors
&
Counters
(230106 Hybrid)

Pulse
Amplifiers
(2301050
Hybrid)

Analyzer

J6
Analog
Electronics

PS RF FB AMUX

+5, ±15 Power
100KHz Clk
300KHz Clk
POR
DA Control
Fil On
EM On
EM Hi
AMUX Ctrl
AMUX Out
RPA n

Index Pulse
Phase Clock
Guard Pulse

J1
Power/Data

Receivers
26C32

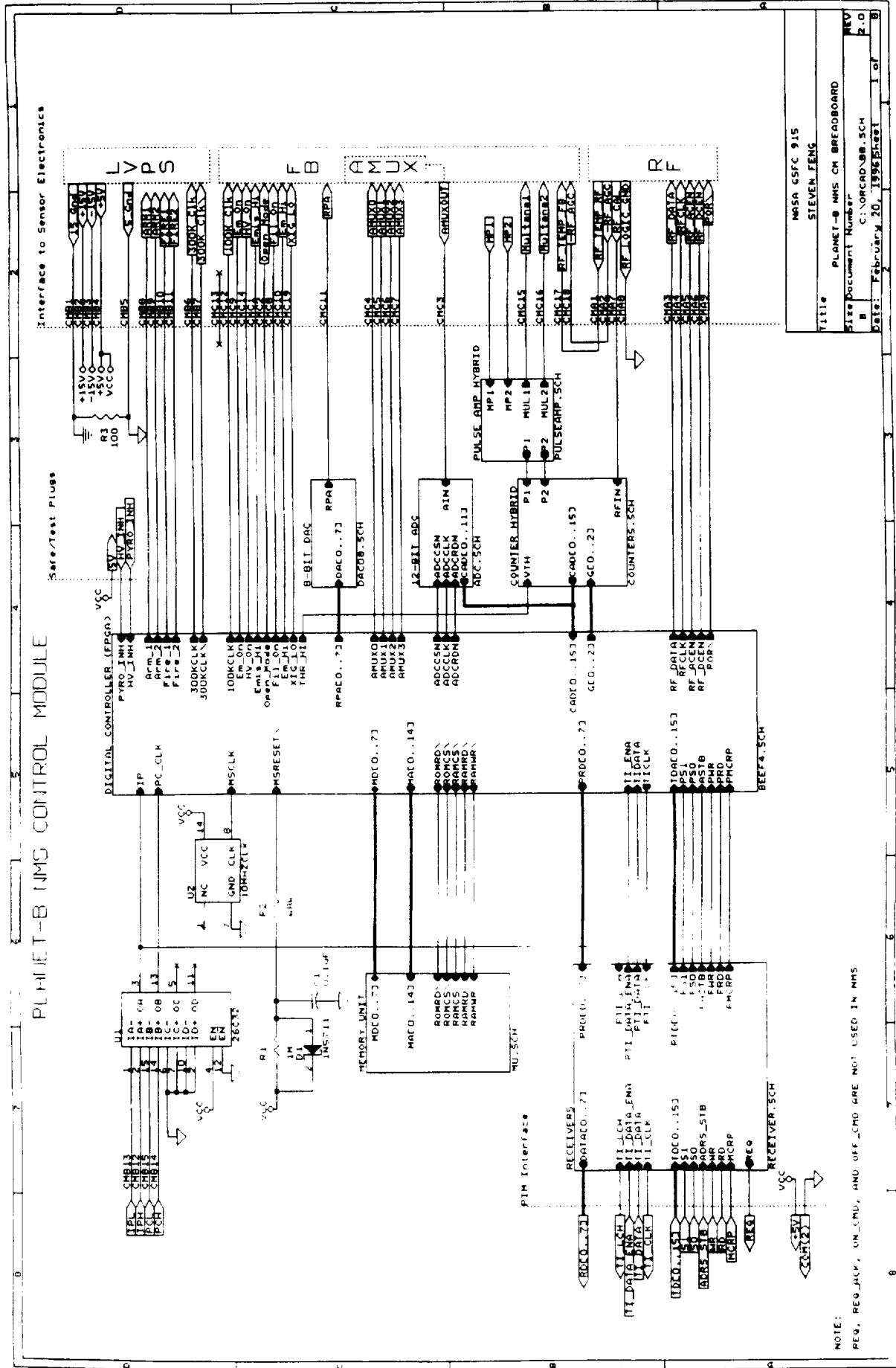
PS
(Passthru)

NASA Goddard Space Flight Center		Control Module Functional Block Diagram Planet - B NMS	
PROJECT 061-NMS	DESIGN J0137A-CM Block	REVISION 061-0137	DATE 12 May 95
DRAWN 6/12/95		CHECKED 6/12/95	
APPROVED 6/12/95		DRAWN 6/12/95	
RELEASE 12 May 95		DRAWN 6/12/95	
PROJECT 061-NMS		DESIGN J0137A-CM Block	
REVISION 061-0137		DATE 12 May 95	
DRAWN 6/12/95		CHECKED 6/12/95	
APPROVED 6/12/95		DRAWN 6/12/95	
RELEASE 12 May 95		DRAWN 6/12/95	

Name	Description	CM Plug	Destination
Harness 4			
RF Temp/RF	RF thermistor temperature from RF to CM as passthru to FB.		
-RF AGC	RF AGC monitor voltage. 0 to -5V. Passthru from RF to FB	CMA-1 C17	RFB-1
RF Data	Serial data for RF amplitude and rod DC.	CMA-2 C18	RFB-11
RF Clk	Clock for RF amplitude data.	CMA-3 C1	RFB-14
RF AC Enable	Enable for rod AC amplitude data.	CMA-4 C2	RFB-15
RF DC Enable	Enable for rod DC amplitude data.	CMA-5 C3	RFB-2
RF Cal	Logic level, RF frequency, to CM counter. Twisted Pair.	CMA-6 C4	RFB-7
RF Logic Gnd	RF digital ground. Twisted Pair.	CMA-7 C5 Tw	RFB-4
RF Logic Gnd	Power on reset not from CM to reset RF serial IF and latch. Signal high true. (A low signal causes reset).	CMA-8 C6 Tw	RFB-6
RF Logic Gnd	RF digital ground.	CMA-9 C7	RFB-12
RF Logic Gnd		CMA-10 C8	RFB-9
Harness 2			
CM 15 Gnd	15 power return for CM	CMB-1 B32	LVB-16
CM +15	+15 power for CM	CMB-2 A32	LVB-7
CM -15	-15 power for CM	CMB-3 C32	LVB-24
CM 5 Gnd	5 power return. 5 and 15 gnds NOT connected in CM	CMB-4 B30	LVB-15
CM +5	+5 power for CM	CMB-5 A30	LVB-6
300K Sync	300KHz sync for main converter. Balanced pulse xfmr load. Twisted Pair.	CMB-6 A1 Tw	LVB-3
300K Sync	300KHz return. Balanced, differential, input. Twisted Pair.	CMB-7 A2 Tw	LVB-13
Arm-1	+5 logic, true, to HOLD, the pyro arming relay #1 on before fire. A SC command of Disarm-1 must be sent to turn off this signal after firing.	CMB-8 A4	LVB-5
Arm-2	+5 logic, true, to HOLD, the pyro arming relay #2 on before fire. A SC command of Disarm-2 must be sent to turn off this signal after firing.	CMB-9 A5	LVB-23
Fire-1	+5 V logic, true, to fire pyro #1. Must be executed AFTER the Arm-1 cmd. SC cmd of Unfire-1 must be sent after the Disarm-1 cmd.	CMB-10 A6	LVB-9
Fire-2	+5 V logic, true, to fire pyro #2. Must be executed AFTER the Arm-2 cmd. SC cmd of Unfire-2 must be sent after the Disarm-2 cmd.	CMB-11 A7	LVB-18
Index Pulse +	Sun detect pulse from S/C. NMS latches clock into TM word for angle of attack reference. Balanced pair.	CMB-12 A8	LVB-1
Index Pulse -	Return wire of balanced pair.	CMB-13 A9	LVB-10
Phase Clock +	Clock of counts between Index Pulses. -8000 between Index Pulses. Balanced Pair.	CMB-14 A11	LVB-11
Phase Clock -	Return wire of balanced pair.	CMB-15 A12	LVB-19
Guard Pulse +	Pulse from Topside Sounder via S/C. Pulse true before output pulse. Not used in NMS as of 1-Jul-95. Balanced pair.	CMB-16 A14	LVB-2
Guard Pulse -	Return wire of balanced pair.	CMB-17 A15	LVB-12
Harness 5			
Emis Hi	+5V logic signal, true, to hold, ionization voltage at hi (70V).	CMC-1 B1	FBA-3
Open Mode	+5V logic signal, true, to hold, open mode (lower, 2 ion focus voltages).	CMC-2 B2	FBA-12
AMUX Out	AMUX output to 8-bit D/A. -040V to +5.08V. Overload (fault) from -15V to +15V.	CMC-3 B3	FBA-19
AMUX Add 0	AMUX address 0. See Pg. 1.08, Dwg. 061-0114, for MUX assignment.	CMC-4 B4	FBA-1
AMUX Add 1	AMUX address 1.	CMC-5 B5	FBA-10
AMUX Add 2	AMUX address 2.	CMC-6 B6	FBA-2
AMUX Add 3	AMUX address 3.	CMC-7 B7	FBA-11
Fil On	+5 V logic signal, true, to hold, filament on.	CMC-8 B8	FBA-13
EM On	+5 V logic signal, true, to hold, EM on.	CMC-9 B9	FBA-4
EM Hi	+5 V logic signal, true, to hold, EM to 200V higher, when EM is ON.	CMC-10 B10	FBA-6
RPA n	Analog voltage, -10V to +10V, from 8-bit D/A, to apply RPA voltage directly to RPA grnd. Rev 3/5/96	CMC-11 B11	FBA-14
100K Clk	100KHz clock for FB AC drive. FPGA drive to UC1709 load.	CMC-12 B20	FBA-25
Spare		CMC-13 B21	FBA-17
HV On	+5 V logic signal, true, to hold, HV Bias Voltages ON. This signal false if Safe/Arm plug, P3 "Inhibit HV", plug is installed.	CMC-14 B12	FBA-5
Mullana 1	EM analog voltage from PA #1 to FB AMUX input.	CMC-15 B14	FBA-22
Mullana 2	EM analog voltage from PA #2 to FB AMUX input.	CMC-16 B15	FBA-23
RF Temp/FB	RF thermistor temperature from RF, to FB, as passthru, from RF.	CMC-17 B17	FBA-20
-RF AGC	RF AGC monitor voltage. 0 to 15V. Passthru to FB.	CMC-18 B18	FBA-21
XIG Low	+5 V logic signal, true, to hold, XIG low (0 volts), by command. False sets XIG at +-45V (normal).	CMC-19 B13	FBA-7
Harness 6			
Plus 5	Plus 5 volts to Safe/Arm connector for pull-up of inhibits.	ST-1 A1	CMD-1
HV Inhibit	When pulled high, HV (both 500 and EMs) are logically held off.	ST-2 B1	CMD-2
Pyro Inhibit	When pulled high, the pyro signals are inhibited	ST-3 C1	CMD-3

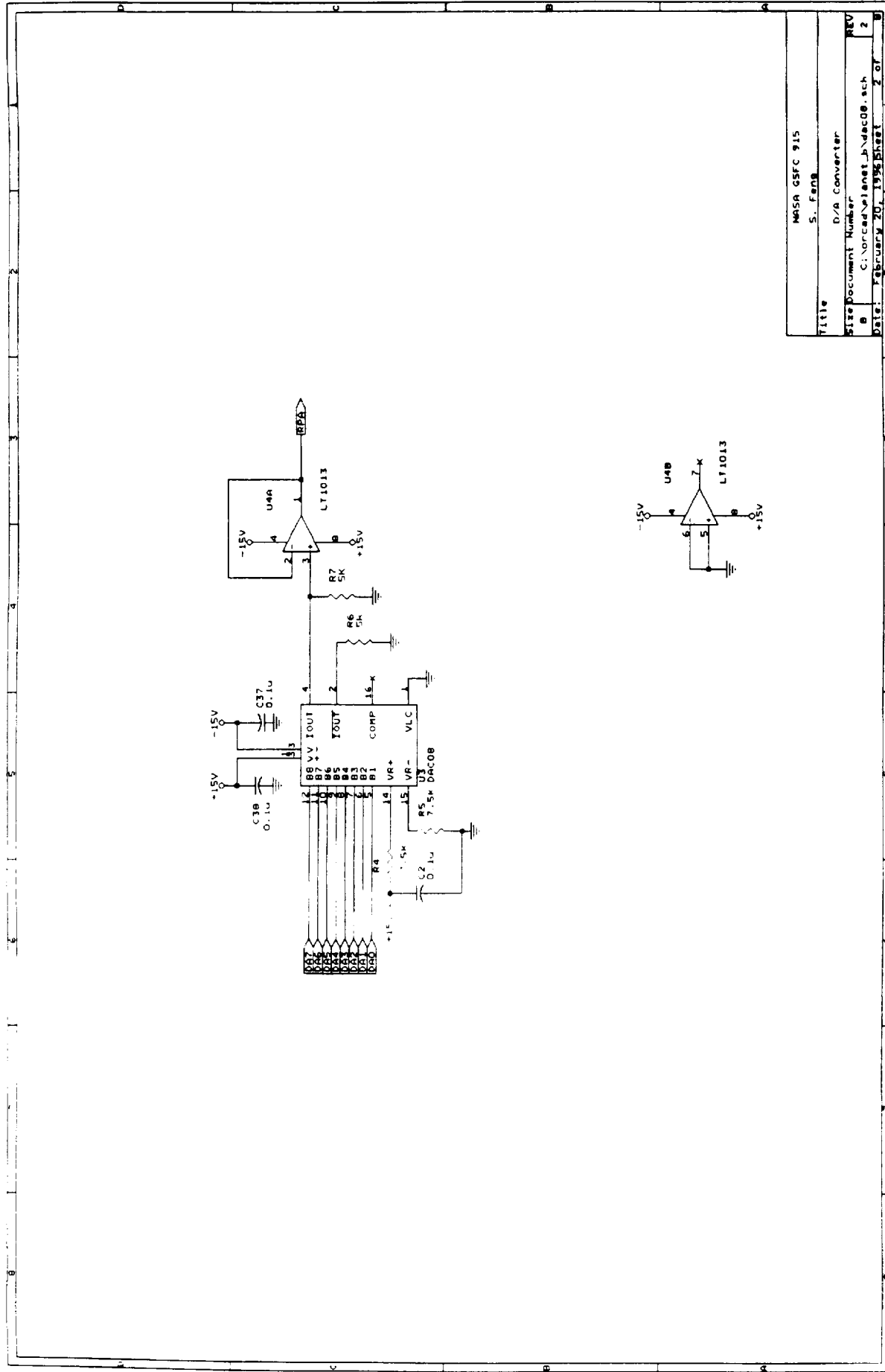
*Note 5/11/95: Pin assignments are placeholders only. Pin numbers will be assigned during PCB layout.

PLANET-B NMS CONTROL MODULE



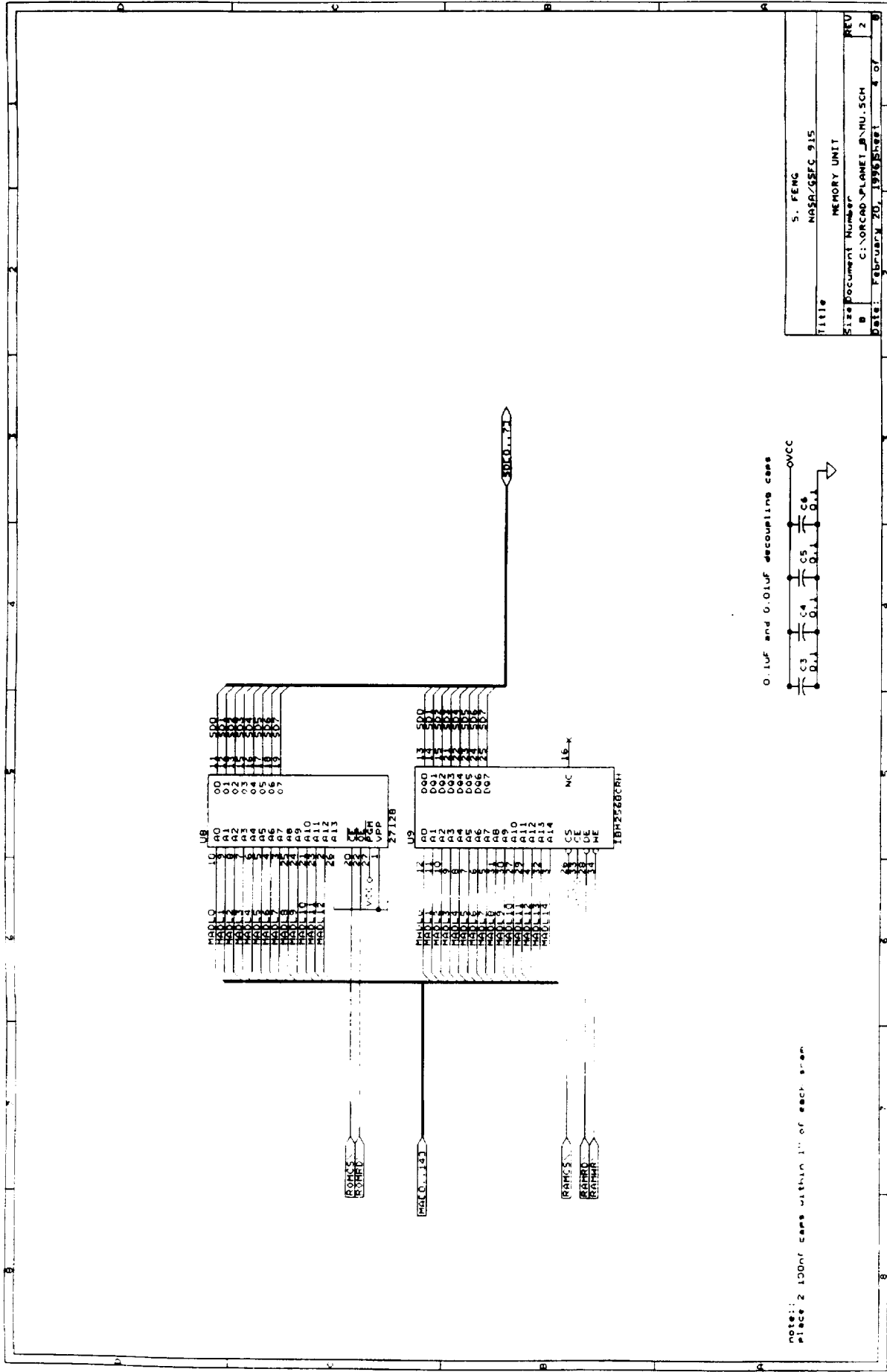
NOTE:
 REG, REG_A, ON_CMD, AND OFF_CMD ARE NOT USED IN NMS

Title		NASA GSFC 915	
Author		STEVEN FENG	
File Name		PLANET-B NMS CH BREADBOARD	
Revision		REV 2.0	
Date		February 20, 1996	
Sheet		1 of 8	



Title	NASA GSFC 915
Author	S. Fong
Size	D/A Converter
Document Number	C:\nasa\planet_b\dac08.sch
Rev	2
Date	February 20, 1995
Sheet	2 of 2

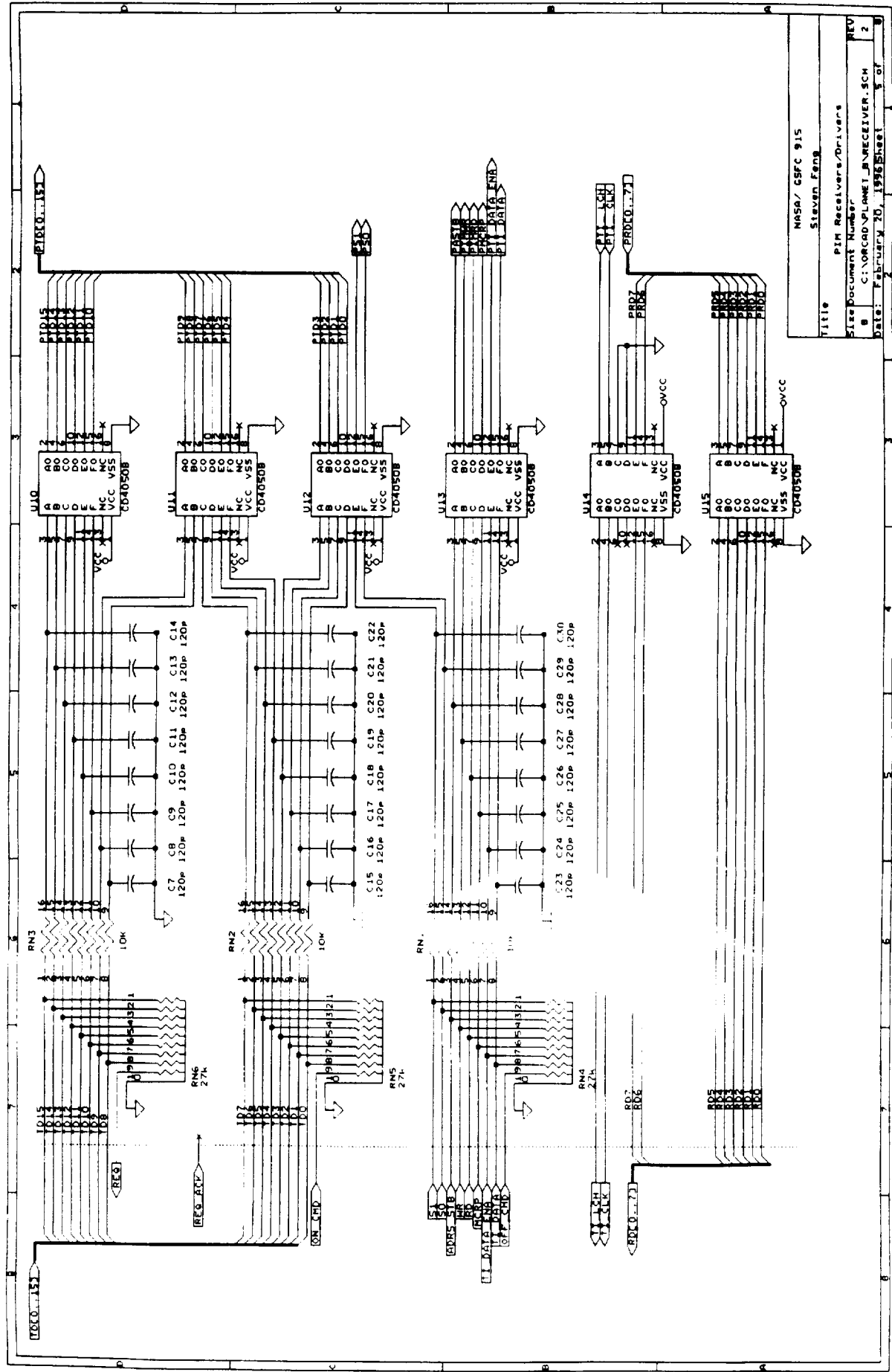
Planet-B NMS Control Module Pg 3 of 8 SPRL File:061-0406



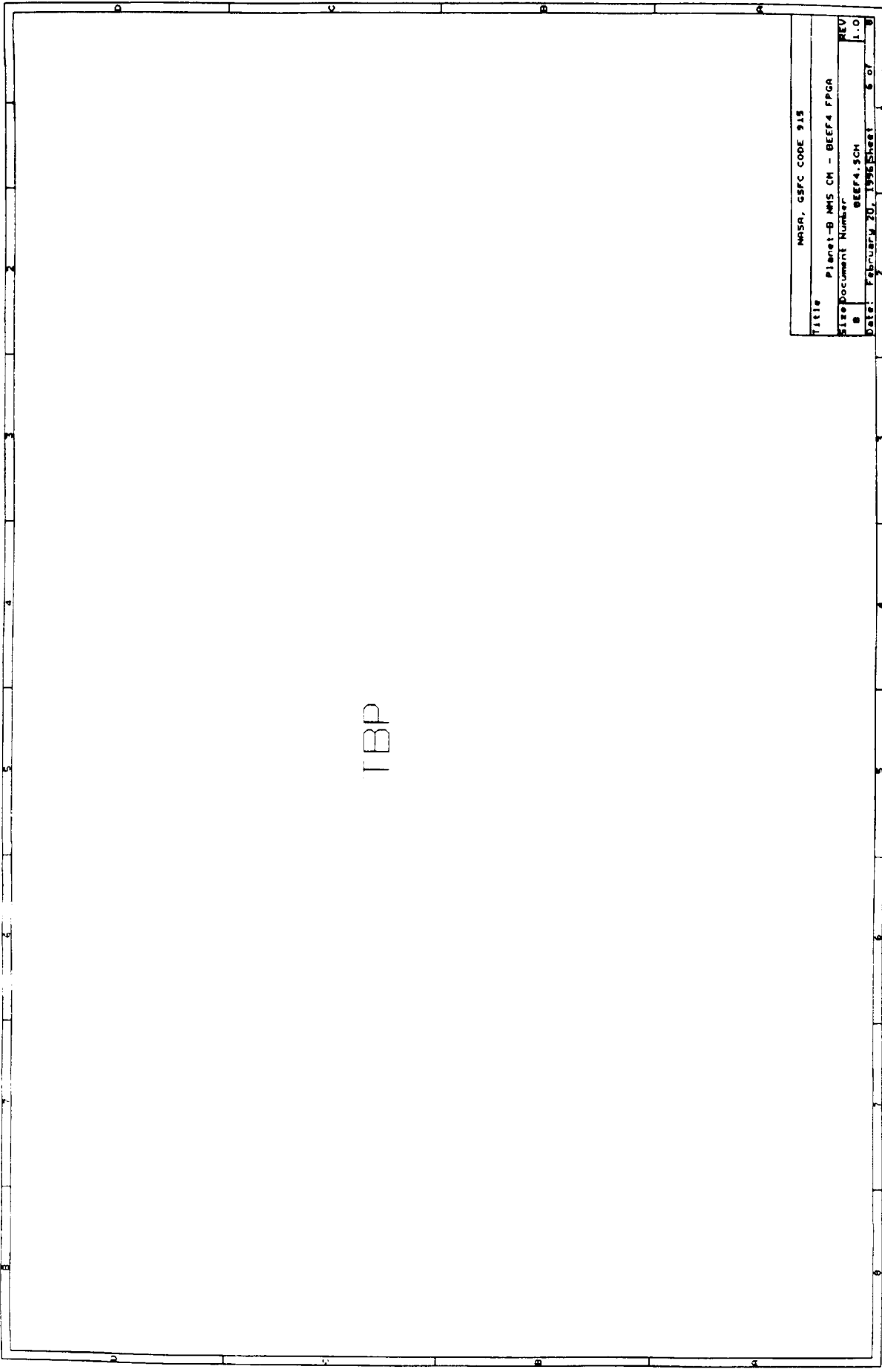
note:
place 2 100nf caps within 1" of each stem

0.1uF and 0.01uF decoupling caps

5. FENG	
NASA/SSFC 915	
MEMORY UNIT	
Size Document Number	
B C:\NORCAD\PLANET_B\NMU.SCH	REV 2
Date: February 20, 1995	Sheet 4 of 8



Title	NASA/ GSFC 915
Size	Pin Receivers/Drivers
Document Number	Size Document Number
REV	2
Date	February 20, 1995
Sheet	5 of 8



NASA, GSFC CODE 915	
Title	Planet-B NMS CH - BEEF4 FPGA
Size Document Number	BEEF4.SCH
REV	1.0
Date	February 20, 1995 Sheet 6 of 8

