ENVIRONMENTAL BARRIER COATINGS HAVING A YSZ TOP COAT

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ABSTRACT

Environmental barrier coatings (EBCs) with a Si bond coat, a yttria-stabilized zirconia (YSZ) top coat, and various intermediate coats were investigated. EBCs were processed by atmospheric pressure plasma spraying. The EBC durability was determined by thermal cycling tests in water vapor at 1300°C and 1400°C, and in air at 1400°C and 1500°C. EBCs with a mullite $(3Al_2O_3 \cdot 2SiO_2) + BSAS$ $(1 - xBaO \cdot xSrO \cdot Al_2O_3 \cdot 2SiO_2)$ intermediate coat were more durable than EBCs with a mullite intermediate coat, while EBCs with a mullite/BSAS duplex intermediate coat resulted in inferior durability. The improvement with a mullite+BSAS intermediate coat was attributed to enhanced compliance of the intermediate coat due to the addition of a low modulus BSAS second phase. Mullite+BSAS/YSZ and BSAS/YSZ interfaces produced a low melting (<1400°C) reaction product, which is expected to degrade the EBC performance by increasing the thermal conductivity. EBCs with a mullite+BSAS/graded mullite+YSZ intermediate coat showed the best durability among the EBCs investigated in this study. This improvement was attributed to diffused CTE mismatch stress and improved chemical stability due to the compositionally graded mullite+YSZ layer.

INTRODUCTION

Silicon-based ceramics, such as silicon carbide fiber reinforced silicon carbide (SiC/SiC) ceramic matrix composites (CMC), are prime candidates for the hot section components of next generation gas turbine engines. Silicon-based ceramics, however, suffer from rapid surface recession in combustion environments due to the volatilization of silica scale via reaction with water vapor.[1-4] An external environmental barrier coating (EBC) is a logical approach to achieving protection and long-term stability.

The early coatings work for Si-based ceramics was initiated to provide protection from contact stress damage or molten salt corrosion. Garrett Turbine Engine Company initiated coatings work in the late 1970s - early 1980s to seek solutions to the contact stress damage on Si-based ceramics.[5,6] Various refractory oxides, such as Al₂O₃, ZrO₂, Cr₂O₃, AlN, and mullite (3Al₂O₃•2SiO₂), were investigated by plasma spraying, chemical vapor deposition (CVD), sol-gel, and electron beamphysical vapor deposition (EB-PVD). In 1980s, Solar Turbines developed plasma-sprayed refractory oxide coatings for hot corrosion protection of SiC ceramics.[7,8] Among the coatings investigated, which included Al₂O₃, YSZ, mullite, zircon (ZrSiO₄), and Y₂O₃, single layer mullite coatings showed the best durability in hot corrosion test at 760 - 1204°C. In the early 1990s, NASA Glenn Research Center (GRC) modified the plasma spray process to deposit fully crystalline mullite coatings that exhibited dramatically improved hot corrosion and thermal cycling resistance compared to conventionally plasmasprayed mullite coatings at temperatures up to 1300°C.[9,10] In the mid 1990s, Boston University developed a mullite coating by CVD [11], which was later modified at ORNL.[12] Thin (2~10 µm), dense CVD mullite coatings provided excellent protection in Na₂SO₄. Other coating systems for Si-based ceramics investigated in 1980s-1990s are summarized in Ref.[13].

By the mid 1990s, the volatilization of silica in water vapor and the resulting surface recession of silicon-based ceramics emerged as a showstopper for the application of these materials

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in combustion environments. This shifted the focus of coatings research to protection from water vapor.[14-16] Mullite coatings also lost silica in water vapor through volatilization, leaving a porous alumina layer on the surface in a simulated combustion environment, while mullite/YSZ duplex coatings effectively suppressed the silica volatilization (Fig. 1).[15] Mullite/YSZ duplex coatings, however, eventually suffered from water vapor attack as water vapor penetrated through cracks in the coating and attacked the substrate.



Figure 1. Specific weight change of coated and uncoated SiC in a high pressure burner rig environment (1230° C, 6 atm, pH₂O ~ 0.6, gas velocity ~ 24 m/s) [15]

A new, crack-resistant coating system, consisting of a silicon bond coat, a mullite or modified mullite (mullite+BSAS) intermediate coat, and a BSAS (1-xBaO•xSrO•Al₂O₃•2SiO₂) top coat, was developed as a part of the HSR-EPM (High Speed Research-Enabling Propulsion Materials) Program in joint research by NASA, GE, and Pratt & Whitney.[17,18] The EBCs developed in the EPM Program were scaled up and applied on SiC/SiC composite combustor liners by Pratt & Whitney in three Solar Turbines Centaur 50S turbine engines under DOE's Ceramic Stationary Gas Turbine Engine Program.[18,19] The combined operation of the three engines has resulted in the accumulation of over 24,000 hours without failure (~1250°C maximum combustor liner temperature), with the engine used by Texaco in Bakersfield, CA, accumulating about 14,000 hours.

In our laboratory, research is underway to develop EBCs capable of 1482°C (2700°F) EBC surface temperature and 166°C (300°F) temperature gradient. Thorough understanding of current EBCs should provide the foundation upon which development of future EBCs can be based. Therefore, a task was undertaken to characterize current EBCs, with the main

focus on determining the upper temperature limit. Areas investigated include environmental/chemical durability, phase stability, and thermal conductivity.[20-22] It was suggested based on chemical/environmental durability and silica volatility that the upper temperature limit of EBCs based on silicon, mullite, and BSAS, for over a thousand hours of life, is ~1400°C.[20] BSAS-silica (thermally grown) reaction produces a low melting (~1300°C) Al₂O₃-SiO₂-BaO-SrO reaction product, leading to EBC delamination, while BSAS is projected to suffer significant recession (20 ~ 60µm after 1000h at 1400°C, 6 atm, $pH_2O = 0.6$, gas velocity = 24m/s). The EBC on Solar Turbines SiC/SiC liners showed substantial BSAS recession in some areas after the 14,000h engine test.[19] Two approaches are taken in our research to develop EBCs with higher temperature capability. One is to apply a low thermal conductivity TBC top coat, such as YSZ, and the other is to develop high melting, low CTE oxide EBCs. This paper discusses the status of the development of EBCs consisting of a Si bond coat, a YSZ top coat, and various intermediate coats.

EXPERIMENTAL

Table I.	EBC systems	investigated	(500 ~	750	μm thick)	
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Bond coat	Intermediate coat	Top coat	
Si	mullite	YSZ	
Si	mullite+BSAS	YSZ	
Si	mullite/BSAS	YSZ	
Si	mullite+BSAS/graded mullite+YSZ	YSZ	

Table I lists EBC systems investigated in this study. EBCs were applied by plasma spraying onto sintered, monolithic SiC (HexoloyTM, Carborundum, Niagara Falls, NY) and melt infiltrated (MI) SiC/SiC composites (GE Power Systems Composites, Newark, Delaware). Only one face of test coupons (2.5 cm x 0.6 cm x 0.15 cm) was coated. Silicon powder was purchased from Atlantic Equipment Engineers (Bergenfield, NJ), mullite powder from Cerac, Inc. (Milwaukee, WI), BSAS powder from H.C. Starck Inc. (Newton, MA) and Praxair Specialty Ceramics (Woodinville, WA), and mullite+YSZ mixtures from Cambridge Microtech (Cambridge, MA).

T	ahle	II	Test	conditions
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Environment	steam (90% H ₂ O- 10% O ₂)	ambient air
Temperature (°C)	1300, 1400	1400, 1500
Cycle frequency (hour)	1	20
Gas velocity (cm/sec)	2.2	0
Total pressure (atm)	1	1
Substrate	MI SiC/SiC	sintered SiC
Silicon bond coat	yes	no

Table II lists test conditions. The steam environment was generated by pumping water into a cyclic rig using a peristaltic pump. The cyclic rig consists of a high purity alumina tube (99.8% purity) and a vertical tube cycling furnace. The pumped water was rapidly vaporized by passing through quartz wool at T>300°C in flowing oxygen. A box furnace was used for the ambient air environment. The use of monolithic SiC and the absence of Si bond coat in the ambient air exposure were to eliminate the melting of silicon (mp = 1416° C) in the 1500° C exposure. Tests in the ambient air were to investigate the EBC chemical stability. The progress of degradation was determined by visual observation, cross-sectional examination using scanning electron microscopy (SEM), and chemical analysis using energy dispersive spectroscopy (EDS). Phase transformations in YSZ were determined by X-ray diffraction.

RESULTS AND DISCUSSION

Si/Mullite/YSZ

EBCs on MI SiC/SiC remained adherent after a 100h exposure at 1300°C in steam. At 1400°C after a 100h exposure in steam, the EBC completely peeled off, typically along the Si/mullite interface or through the Si bond coat, and subsequently curled up toward the YSZ side. The direction of the curling indicates that the YSZ coating shrunk more than the mullite coating during cooling. The larger shrinkage of the YSZ coating is attributed to YSZ sintering. The sintering of plasma-sprayed YSZ on SiC during thermal exposures is reported in Ref. [23].

The YSZ shrinkage and the YSZ-mullite CTE mismatch (~10 x 10⁻⁶/°C for YSZ vs. ~5 x 10⁻⁶/°C for mullite[24]) are expected to generate in-plane tensile stress in the YSZ layer. causing cracking through the thickness of YSZ. YSZ phase stability is another issue for long-term durability. A nonequilibrium tetragonal phase forms when partially stabilized zirconia-yttria is plasma-sprayed.[25,26] This phase decomposes slowly at high temperatures into a high-yttria cubic phase and a low-yttria tetragonal phase. This equilibrium tetragonal phase transforms on cooling to the monoclinic phase. According to X-ray diffraction, the YSZ top coat lost most of the nonequilibrium tetragonal phase after as early as 20h at 1300°C. This results in a reversible phase transformation between the tetragonal phase and the monoclinic phase on thermal cycling, adding additional stresses on the EBC.

Mullite EBCs fail due to enhanced oxidation via water vapor that penetrates through cracks in the mullite and oxidizes the substrate.[15] Adding a YSZ top coat does not prevent the penetration of water vapor, due to cracking in the YSZ. Instead, the YSZ top coat accelerates the EBC spallation by adding more stresses.

Si/Mullite+BSAS/YSZ

EBCs on MI SiC/SiC remained adherent after a 100h exposure at 1300°C in steam. At 1400°C after 240h in steam, a partial spallation was observed, typically along the Si/mullite+BSAS interface or through the Si bond coat. This is

an improvement compared to the Si/mullite/YSZ EBC, which completely spalled after a 100h exposure at 1400°C in steam. The mullite+BSAS coating possesses superior crack resistance compared to the mullite coating.[17] The young's modulus of mullite (~150 GPa [27]) is larger than that of BSAS (~100 GPa [28]) by a factor of about 1.5. Assuming that plasma-sprayed coatings of mullite and BSAS will maintain a similar modulus ratio, it is proposed that the improved crack resistance of the mullite+BSAS coating is the result of improved compliance due to the addition of a low modulus BSAS phase. Significant cracking was also observed, typically through the thickness of YSZ and along the mullite+BSAS/YSZ interface (Fig. 2).



Figure 2 Cross-section of the Si/mullite+BSAS/YSZ EBC on MI SiC/SiC after 240h at 1400°C in steam.

A significant BSAS-silica reaction was observed at the Si/mullite+BSAS interface, especially on delaminated or spalled areas. This indicates that oxidation of silicon and the subsequent BSAS-silica reaction contributed to the delamination and spalling of the EBC. It is proposed that the combination of EBC stresses and the environmental/chemical degradation at the Si/mullite+BSAS interface is mainly responsible for the EBC spallation. The bright phase in the mullite+BSAS intermediate coat in Fig. 2 is BSAS, elongated in the y direction due to the splat formation during the deposition, and the dark phase is mullite. The morphology of the EBC did not change much from the as-processed EBC, except for the Si/mullite+BSAS interface, which became fuzzy due to the BSAS-silica reaction. In some areas, pores in the YSZ top coat were filled with a reaction product, which appears to have been molten at 1400°C, due to a reaction at the mullite+BSAS/YSZ interface.

The (mullite+BSAS)-YSZ reaction became more pronounced at 1500°C. Figure 3 shows the mullite+BSAS/YSZ interface after 20h at 1500°C in air. Note the changes in the morphology of the intermediate coat compared to the EBC in Fig. 2: the bright phase became smaller in size and more closely spaced. Also note that pores in the YSZ are completely filled with a reaction product. EDS analysis detected only aluminum from the dark phase (area A in Fig. 3), and aluminum, silicon, barium (strontium peak overlaps with the silicon peak), yttrium, and calcium from the bright phase (area B in Fig. 3). It appears that silica from the mullite and yttria from the YSZ reacted with BSAS, forming the molten reaction product. The calcium is speculated to come from impurities in powders or from insulation refractory materials of the box furnace.





Figure 3 Cross-section and EDS spectrum from the Si/mullite+BSAS/YSZ EBC on MI SiC/SiC after 20h at 1500°C in air



Figure 4 Y₂O₃-Al₂O₃-SiO₂ phase diagram [29]

According to the Y₂O₃-Al₂O₃-SiO₂ phase diagram (Fig. 4), the ternary system is capable of forming compounds with the melting point as low as ~1400°C even without BaO, SrO, and CaO.[29] The presence of these alkali earth metal oxides will further reduce the melting point of reaction products in the ternary system. A molten reaction product is expected to increase EBC thermal conductivity because it tends to fill open spaces in the EBC. The thermal conductivity of BSAS-coated MI SiC/SiC continuously increased during a thermal exposure, which was attributed to the formation of a low melting reaction product by the BSAS-silica reaction.[22] The reaction product moved to the YSZ surface forming a molten surface layer, which is expected to be swept away by a high velocity combustion gas. Therefore, the mullite+BSAS/YSZ interface should be avoided to prevent the formation of the low melting reaction product.

Si/Mullite/BSAS/YSZ

EBCs were completely spalled along the BSAS/YSZ interface after a 100h exposure at 1300°C and 1400°C in steam. This suggests a weak BSAS-YSZ adherence. At 1400°C after 120h in air, the EBC was partially delaminated along the BSAS/YSZ interface. Significant reactions were observed at the BSAS/YSZ and the mullite/BSAS interfaces. Figure 5 shows the BSAS/YSZ and the mullite/BSAS interfaces after 120h at 1400°C in air.



Figure 5 Cross-section and EDS spectrum from (a) BSAS/YSZ and (b) mullite/BSAS interfaces after 120h at 1400°C in air

The EDS spectrum from the BSAS/YSZ reaction zone (Fig. 5a) is similar to that of the reaction product at the Similar mullite+BSAS/YSZ interface. to the Si/mullite+BSAS/YSZ EBC, pores in the YSZ were filled with the reaction product, indicating that the reaction product was molten at 1400°C. The EDS spectrum from the mullite/BSAS reaction zone (Fig. 5b) contains aluminum, silicon, and barium. The intensity of these peaks is between that of mullite and BSAS peaks, suggesting that the reaction zone is the result of interdiffusion between the two layers. There were no indications that the reaction zone was molten at 1400°C. The mullite-BSAS reaction appears to be benign to EBC durability at least for the short-term exposure conducted in this study. Significant diffusion between mullite and BSAS was also observed in the Si/mullite/BSAS EBC after a 1000h exposure at 1300°C in steam, with no obvious detrimental effects on EBC durability.[20]

Si/Mullite+BSAS/Graded Mullite+YSZ/YSZ

The mullite/YSZ interface maintains excellent chemical stability at temperatures up to 1500°C, suggesting that mullite can be an effective chemical barrier between mullite+BSAS and YSZ. One way to alleviating the CTE mismatch stress between the YSZ and the mullite+BSAS coatings is by creating a compositionally graded intermediate coat, starting with mullite at the bottom then gradually increasing the YSZ/mullite ratio towards the YSZ top coat. A graded layer of mullite/mullite+25v/o YSZ/mullite+50v/o YSZ/mullite+75v/o YSZ was applied between the mullite+BSAS and the YSZ coatings. Figure 6 shows the cross-section of Si/mullite+BSAS/graded mullite+YSZ/YSZ EBC (~750µm thick) on MI SiC/SiC after 370h at 1400°C in steam. The EBC showed a partial spallation after the exposure, an improvement compared to the Si/mullite+BSAS/YSZ EBC, in which a partial spallation was observed after 240h at 1400°C in steam.



Figure 6 Cross-section of the Si/mullite+BSAS/graded mullite+YSZ/YSZ EBC on MI SiC/SiC after 370h at 1400°C in steam

The failure mode of the EBC was more complex than the other EBCs. Although the spallation occurred typically along the Si/mullite+BSAS interface or through the Si bond coat, some areas spalled through the mullite+BSAS or the mullite layers. Significant cracking was also observed, typically through the thickness of graded mullite+YSZ/YSZ and along or near the mullite+BSAS/mullite interface as shown in Fig. 6. There were no indications of low melting reaction products within the EBC, except for the BSAS-silica reaction product at the Si/mullite+BSAS interface. The superior durability of the Si/mullite+BSAS/graded mullite+YSZ/YSZ EBC compared to the other EBCs investigated in this study is attributed to the improved chemical stability and the diffused CTE mismatch stress due to the compositionally graded mullite+YSZ layer.

SUMMARY

EBCs with a Si bond coat, a YSZ top coat, and various intermediate coats were applied by plasma spraying onto SiC ceramics. Coated coupons were tested in steam and in ambient air to investigate the EBC durability.

The Si/mullite/YSZ EBC completely spalled along the Si/mullite interface or through the Si bond coat after a 100h exposure at 1400°C in steam. The Si/mullite+BSAS/YSZ EBC showed a partial spallation after 240h at 1400°C, indicating improved EBC durability compared to the Si/mullite/YSZ EBC. The mullite+BSAS/YSZ interface, however, produced a low melting reaction product (<1400°C), which filled pores in the YSZ. A low melting reaction product is detrimental to EBC performance by increasing the thermal conductivity. Further improvement in EBC durability was achieved with a mullite+BSAS/graded mullite+YSZ intermediate coat, which showed a partial spallation after 370h at 1400°C in steam. The improvement with the mullite+BSAS intermediate coat is attributed to the enhanced compliance of the intermediate coat due to the addition of a low modulus BSAS second phase, while the additional improvement with the mullite+BSAS/graded mullite+YSZ intermediate coat is attributed to diffused CTE mismatch stress and improved chemical stability due to the graded mullite+YSZ layer.

Replacing the mullite intermediate coat with a mullite/BSAS duplex layer was not beneficial to EBC durability. Not only the BSAS/YSZ interface was a "weak link", leading to a complete EBC spallation along the interface after a 100h exposure at 1300°C and 1400°C in steam, but also the BSAS/YSZ interface produced a low melting reaction product similar to that formed at the mullite+BSAS/YSZ interface.

The combination of a "weak link", a weak interface in the EBC caused by a defect or degradation, and the stress caused by YSZ sintering, CTE mismatch, and YSZ phase instability is mainly responsible for the failure of EBCs investigated in this study. The location of the "weak link" depends on the design of

intermediate coat. The Si/mullite interface is the "weak link" for the Si/mullite/YSZ EBC, and the Si/mullite+BSAS interface for the Si/mullite+BSAS/YSZ and the Si/mullite+BSAS/graded mullite+YSZ/YSZ EBCs. The "weak link" along the Si/mullite interface is due to oxidation of the Si bond coat, while the "weak link" along the Si/mullite+BSAS interface is due to the combination of Si bond coat oxidation and the subsequent BSAS-silica reaction. The BSAS/YSZ interface is the "weak link" for the Si/mullite/BSAS/YSZ EBC, which is due to the weak BSAS-YSZ adherence.

A complex thermal/environmental barrier coating system for silicon-containing materials, consisting of a silicon bond coat (optional), a multilayer intermediate coat and a YSZ TBC top coat, is described in Ref. [30]. The multilayer intermediate coat consists of a mullite-containing layer, an alkaline earth metal aluminosilicate (preferably BSAS) layer, and a YSZcontaining layer (discrete or compositionally graded) that also contains mullite, alumina, and/or alkaline earth metal aluminosilicate (preferably BSAS). No durability data were reported in the reference, however, in light of the findings in this study, chemical reactions between the alkaline earth metal aluminosilicate component and YSZ in the multilayer intermediate coat are likely to become a durability issue.

It should be noted that 1400°C is too high for the EBC/substrate interface of current EBC systems, causing the detrimental BSAS-silica reaction. If the EBC/substrate temperature is kept at or below 1316°C (current EBC/substrate temperature goal) the BSAS-silica reaction will be significantly reduced, presumably resulting in an extended EBC life.

CONCLUSIONS

Current EBCs fail by delamination and spallation along a "weak link". The stress caused by a YSZ layer accelerates the failure. A key source for the creation of the "weak link" is environmental/chemical degradation. This proposed failure mechanism suggests that key material properties for long life EBCs should include environmental/chemical stability, low CTE, low modulus, sinter resistance, low thermal conductivity, and phase stability.

A test facility capable of producing a heat flux high enough to generate necessary temperature gradients as well as thermal cycling and steam environments is needed for more efficient identification of promising EBCs.

ACKNOWLEDGEMENTS

I am grateful to G. W. Leissler of QSS/NASA GRC for the preparation of plasma spray coatings, S. L. Leissler of QSS/NASA GRC for the metallographic preparation of EBCs, and R. Garlick of NASA GRC for X-ray diffraction. I am also grateful to J. I. Eldridge of NSAS GRC and D. Zhu of OAI/NASA GRC for many helpful discussions. This work was supported by the NASA Ultra Efficient Engine Technology (UEET) Program.

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