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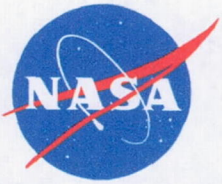
# Overview of The High Performance Antiproton Trap (HiPAT) Experiment

**NASA/Marshall Space Flight Center  
Propulsion Research Center TD40  
Huntsville, Alabama**

**11-14-2002**

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Raymond Lewis – RLewis Company  
Wallace Fant – Cortez III**



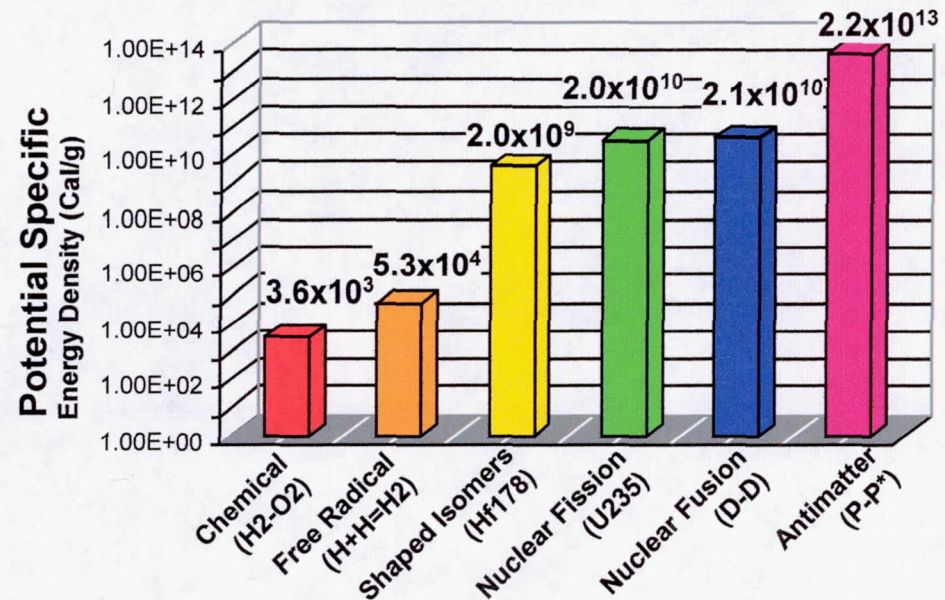


# Why Antimatter??

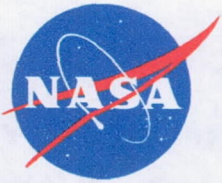
The annihilation of antiprotons with protons represents the highest energy density of any known reaction  $10^8$  MJ/g: the ultimate form of stored energy for future high specific impulse deep space missions.

- 42 mg of antiprotons = energy of 750,000 kg fuel/oxidizer on the Space Shuttle ET
- Envisioned antimatter initiated propulsion concepts require 0.1 to 10 micrograms of antiprotons.
- Storage is a key enabling technology required by all users of antiprotons (NASA and commercial).
- Current production sufficient to evaluate basic handling/utilization technologies.

## Available Energy Sources







# HiPAT Applicability

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Matter/Antimatter annihilation represents the “ultimate” source of stored energy for space propulsion

- The potential benefits to propulsion suggest a phased low level research program.
- Research activities focused on the basic technologies are required to assess its potential.
- Existing antiproton production facilities provide levels sufficient for proof of concept research.
- Results of these assessments can be used to determine further investment.

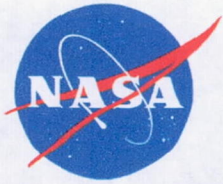
Antiproton storage is a fundamental technology required to experimentally assess utilization methods. The HiPAT device provides a critical resource to the research community supporting basic evaluation

- Knowledge in the operations required for the basic handling and manipulation of antiprotons.
- Development of techniques and basic insight into the operation at production facilities.
- Provides an accumulator enabling single shot experimental testing of propulsion concepts.
- Serves as a front end to research related to high density storage of antimatter.

The HiPAT provides an asset to commercial based enterprises

- Support of research in the medical field related to the development of radio isotopes production and tumor treatment techniques.



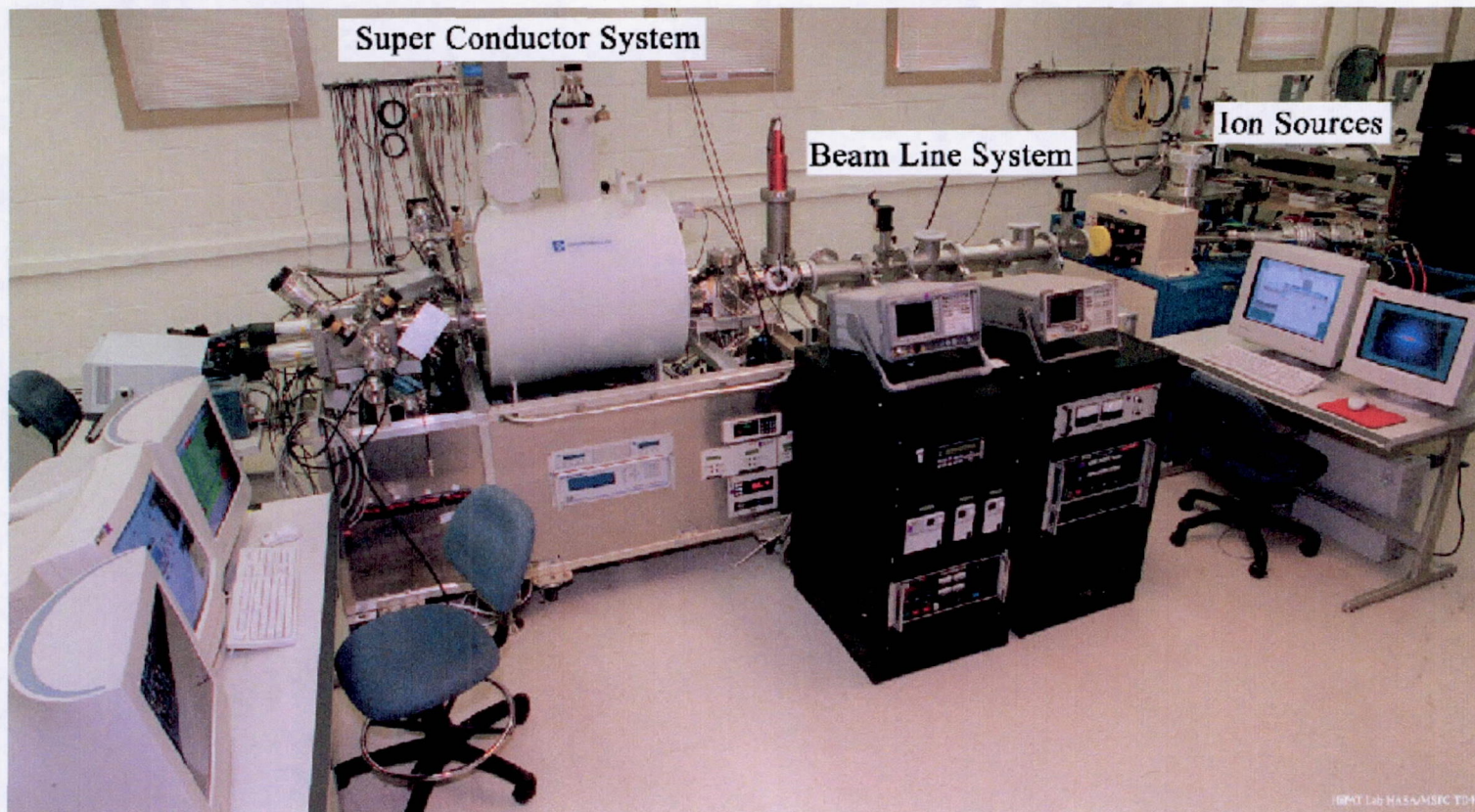


# Approach - Goals

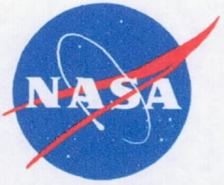
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To address the storage issue, a test device termed the High Performance Antiproton Trap (HiPAT) has been designed and fabricated.

- Electromagnetic Penning-Malmberg design
- Capacity of up to  $1 \times 10^{12}$  antiprotons
- Storage lifetimes of 18 days or more
- Ultra high vacuum system ( $< 10^{-11}$  torr)
- Capable of portable operation
- RF stabilization and passive particle detection





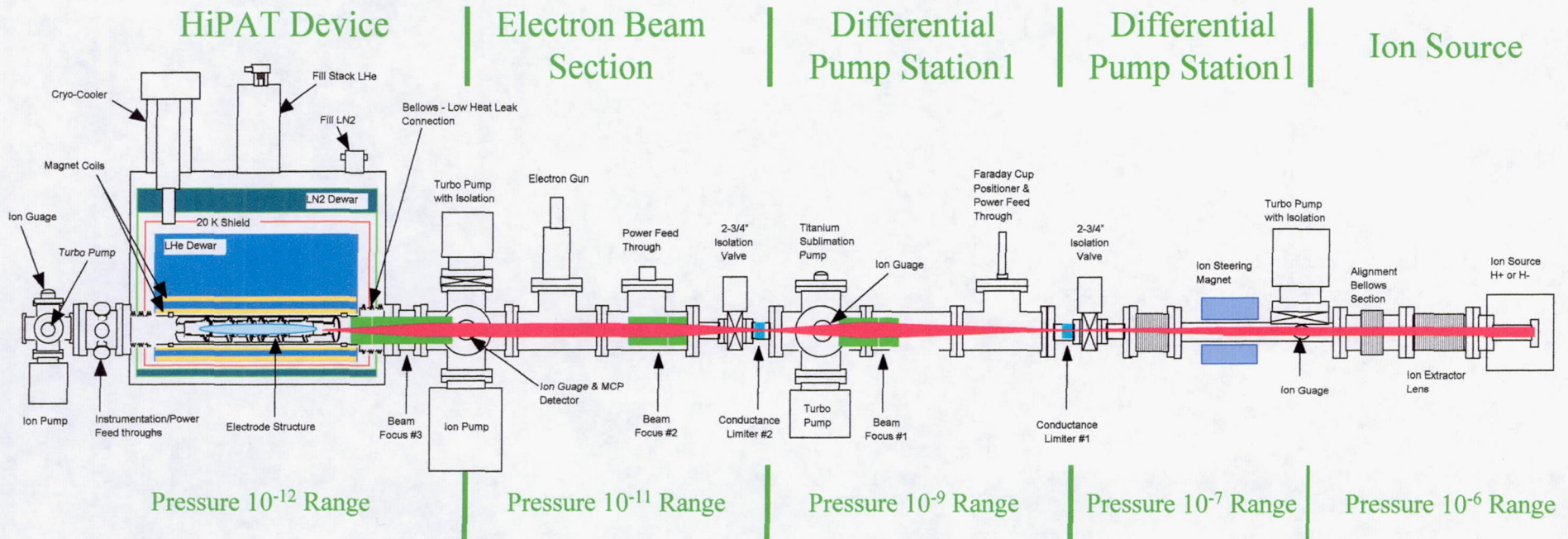


# HiPAT General Layout

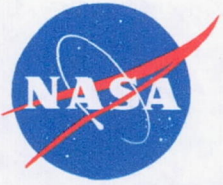
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## Beam Line, Ion Source, and Superconducting Magnet Hardware

- Designed around an ultra high vacuum system with differential pumping capability (maintains 6 orders of magnitude between trap and ion sources).
- Vacuum level ( $10^{-12}$  torr range) reduces loss by radial diffusion and annihilation.
- LHe/LN<sub>2</sub> cooled 4 Tesla superconducting magnet system (end compensated solenoid).
- Hydrogen ion source and hot filament electron gun provide “normal matter” ions.
- High voltage electrostatic beam optics (Einzel lens) to guide and focus ion beams.







# Sizing For Containment

The containment zone — located in the bore of the superconductor — is surrounded by a series of electrodes and insulator segments. The  $10^{12}$  particles are confined radially by the magnetic field and axially by the electric field.

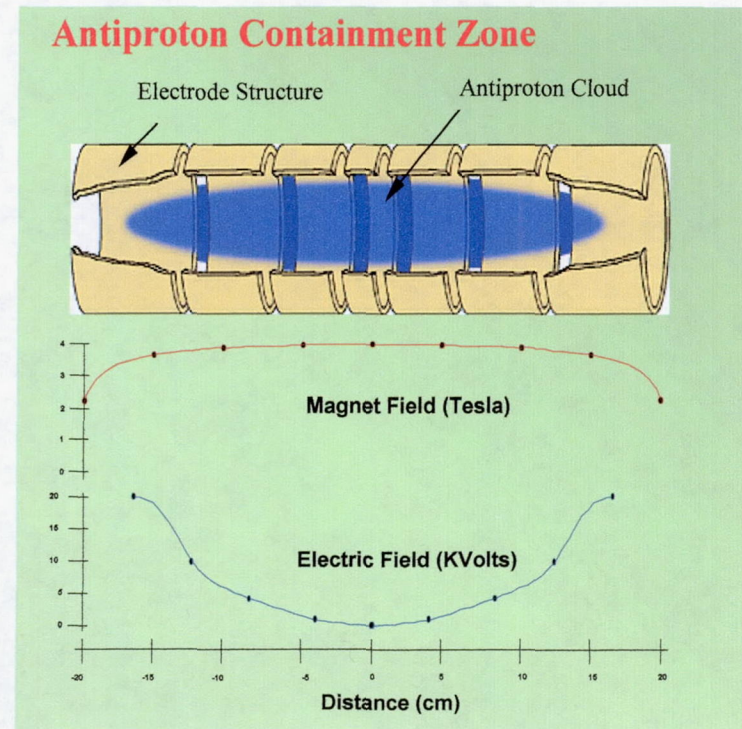
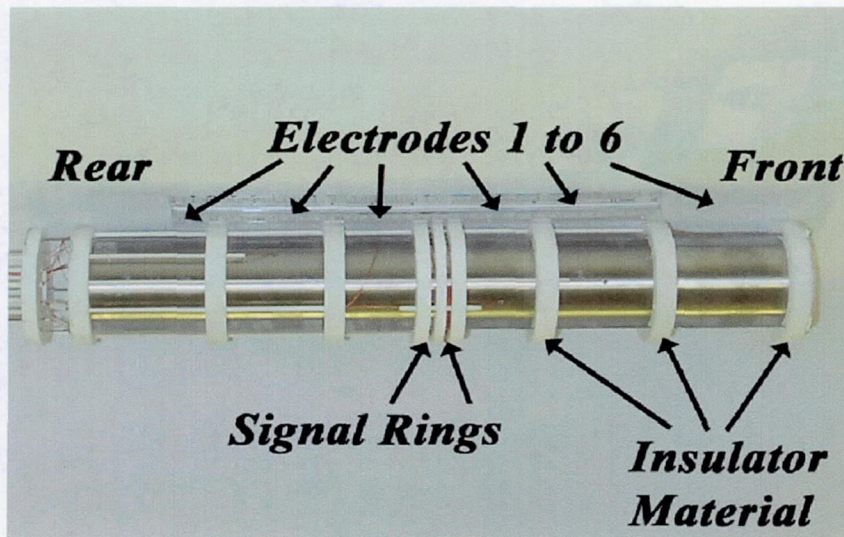
- Magnetic field of 1 Tesla required to balance cloud's radial space charge.
- Electric field of 20 kV required to balance the cloud's axial potential.

Magnetic

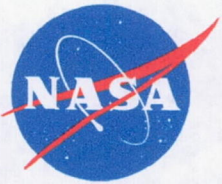
$$N_{\text{Brillouin}} = \frac{B^2}{2\mu_0 M c^2}$$

Electrostatic

$$V_{\text{radius}} = \frac{nq}{4\pi\epsilon_0 R}$$







# Laboratory Operations

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- **Cleaning techniques on the UHV system**
  - Hydrogen glow discharge cleaning (GDC), Titanium sublimation pumps (TSP).
  - Achieve very low vacuum to minimize diffusion loss and ion chemistry/charge exchange
- **Ion production within the containment volume via beam ionization**
  - Simplistic operation using electron and ion beams to generate ions in place.
  - No cycling of electric fields required.
- **Dynamic Capture of externally produced ions**
  - Precision timing of beam line valves, focusing lens and trap electrodes.
  - More closely simulates anticipated operation at antiproton production site.
- **Radio Frequency Systems.**
  - Development of and experiments with particle detection and stabilization techniques.





# Vacuum System Cleaning

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Effort focused on reducing “contaminants” in the vacuum system (e.g., carbon compounds et.al.).

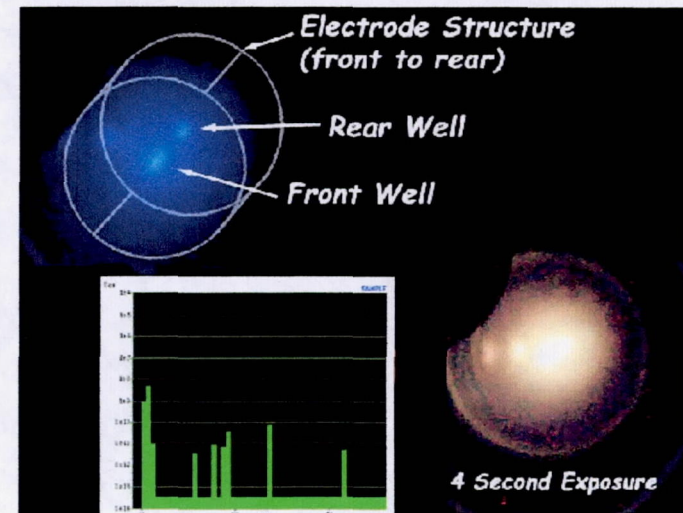
- Minimize charge exchange, preserve hydrogen.
- Increase maximum operating voltage (because of reducing potential for spontaneous glow discharge).

Hydrogen glow discharge techniques to scrub vacuum system.

- DC power up to 500 watts.
- RF power up to 100 watts
- Thermal bake out average 250 °C

## Result

- Current pressure  $7.2 \times 10^{-12}$  torr — **factor of 20 improvement over previous tests.**
- Glow discharge threshold raised from 2 to 10 kV: **visible glow virtually eliminated up to 20 kV.**
- Atm to  $10^{-12}$  torr — **less than a week.**



Spontaneous Glow Discharge



H<sub>2</sub> Glow Discharge Cleaning

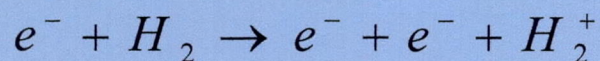
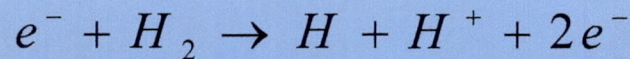




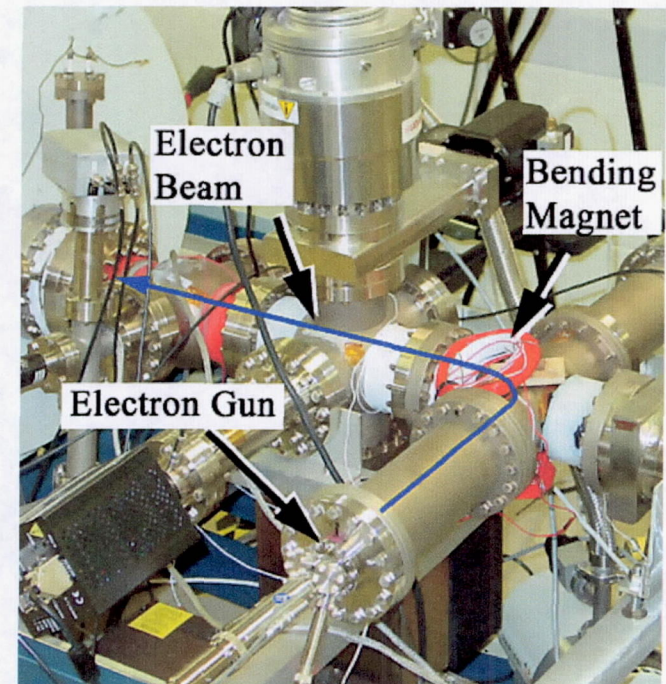
# Ion Production Via Electron Gun

Simple ionization techniques provide straightforward mechanisms to investigate lifetimes and assess RF systems. An electron (or ion) beam can produce trappable ions *in situ*.

- Technique can be called a “poor man’s” ion source
- Primary beam plows through the potential well, ionizing residual background gas (primarily H<sub>2</sub>)
- Energetic (secondary/tertiary/...) electrons and ions also ionize background gas
- Total = primary<sub>ionization</sub> + (e<sup>-</sup> & ions)<sub>2nd</sub> + (e<sup>-</sup> & ions)<sub>3rd</sub> + ...



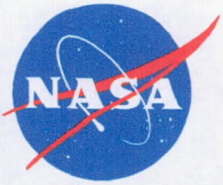
- Probability of formation based on:
  - Background density: n (~1x10<sup>6</sup> /cc)
  - Cross section: σ (~1x10<sup>-16</sup> cm<sup>2</sup>)
  - Path length L (~25 cm) (single pass)



$$e_{total} = \frac{I_{gun} t_{gun}}{1.6 \times 10^{-19}}$$

$$Ions_{total} = n e_{total} \sigma L_{path}$$



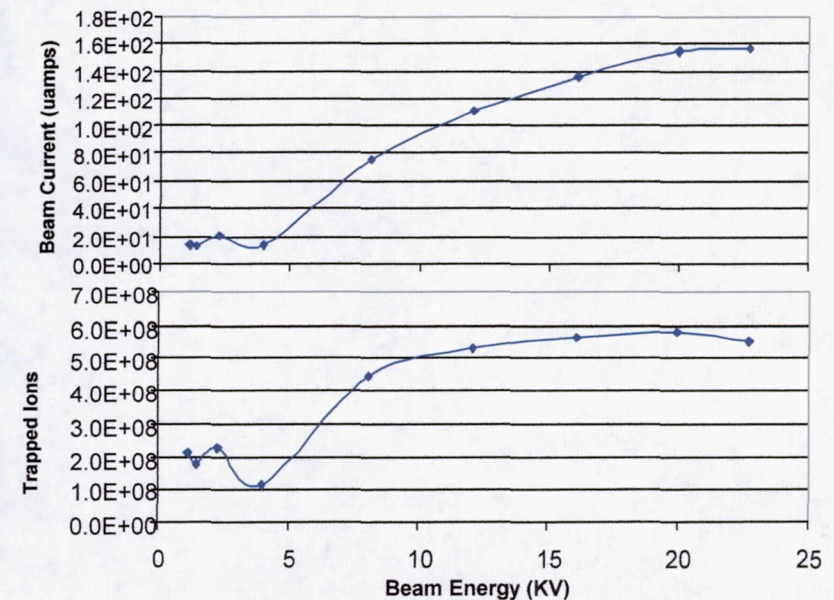
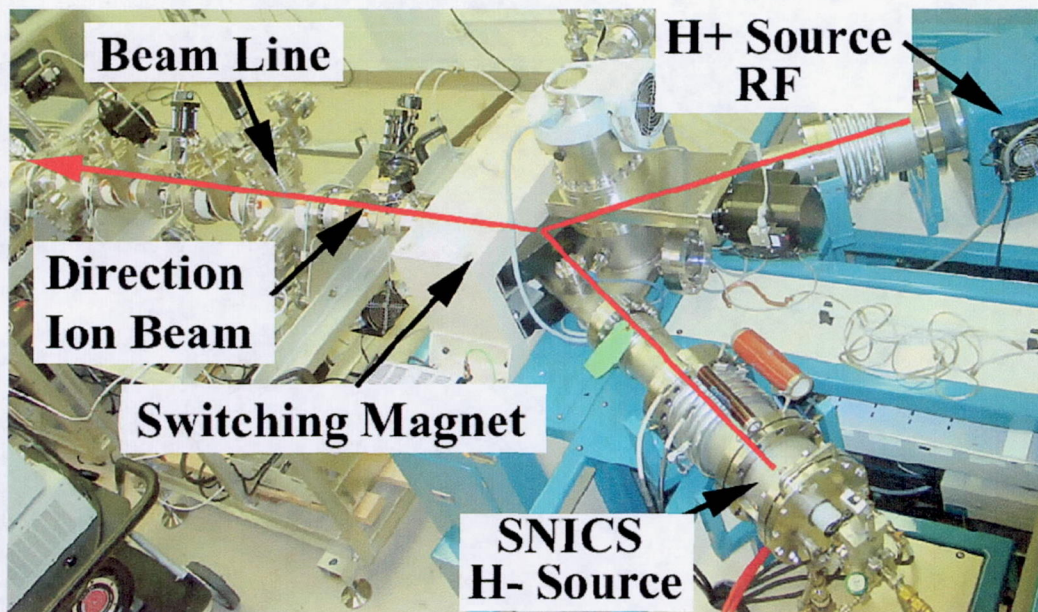


# Particle Capture Via Ion Sources

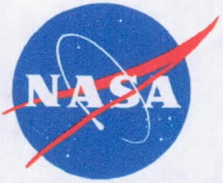
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Demonstrate Quantity and Lifetime of Trapped Ions Using “Normal Matter” Hydrogen Ions ( $H^+$ ) to Simulate Antiprotons.

- NEC source system for ion generation to more closely simulate actual antiproton loading technique.
- Single species ions created externally and transported along beam line to the trap system.
- Source large neutral gas loads require dynamic cycling of isolation valves and differential pumping.





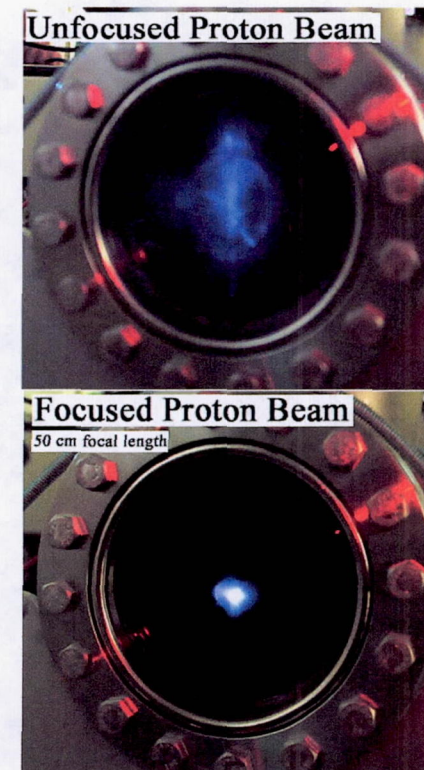
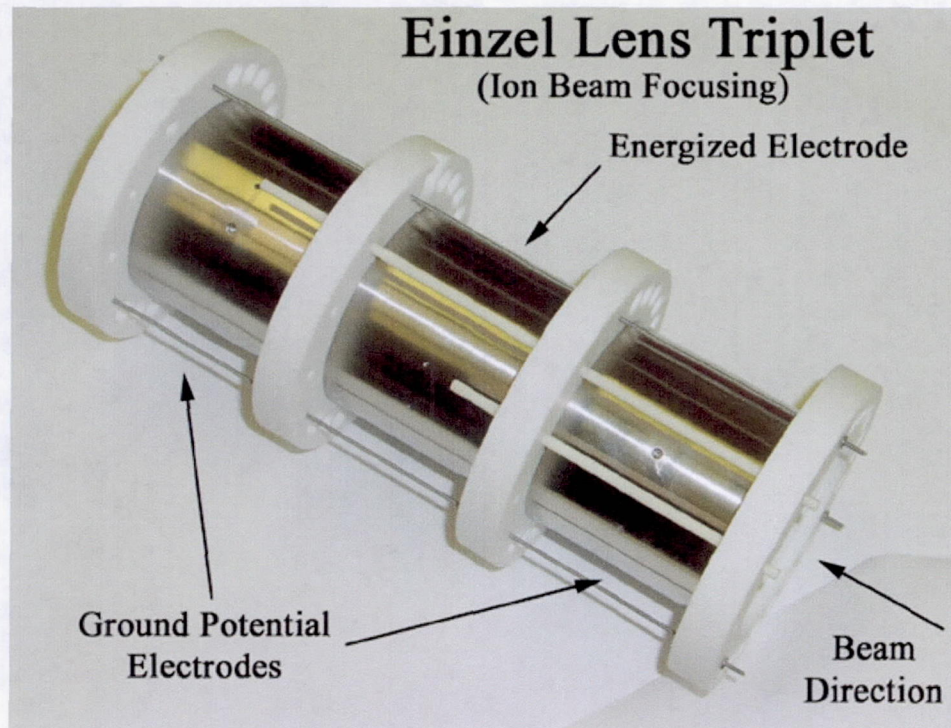


# Ion Beam Steering/Focusing

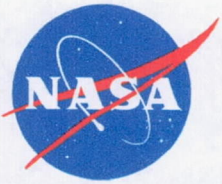
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Transport ion beams from the NEC RF/SNICS sources to the trap system

- Distance of approximately 3 meters requires use of Einzel electrostatic focusing lens
- Two beam line apertures <1cm diameter (differential pumping)
- Compensation against the earth's magnetic field (0.5 gauss)
- Focus to align ions with magnet's fringe field (maximize particle acceptance)
- Movable beam detectors used to fine tune voltages on Einzel lens



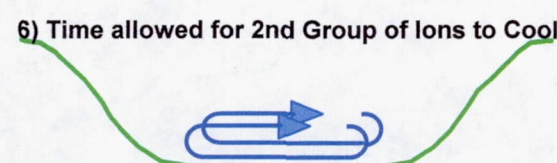
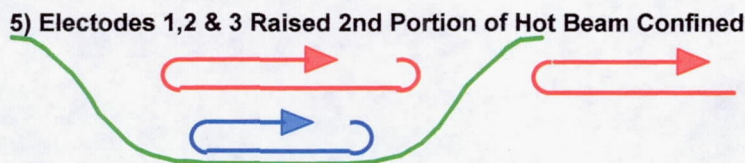
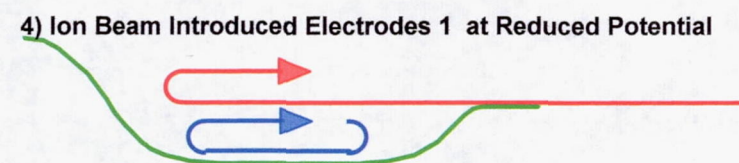
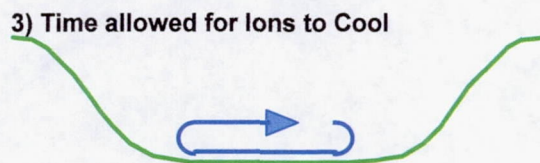
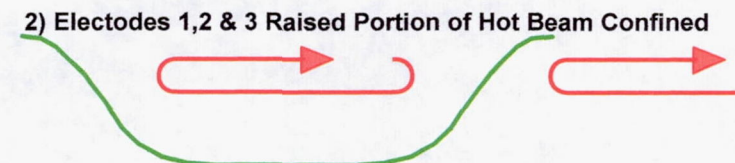
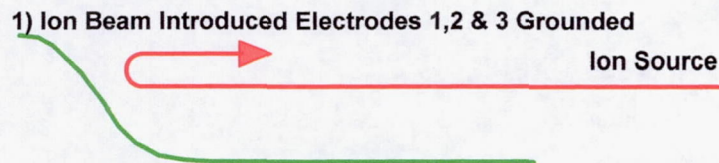
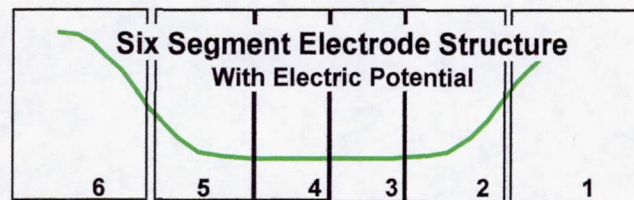




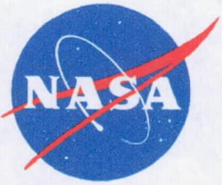
# Ideal Ion Stacking Sequence

Only a limited number of ions can be captured from a single beam spill.  
Reaching higher fill levels necessitates stacking, which entails the following:

- Rapid cycling of electrode groups between a full and reduced electric field condition
- Time must be allowed for hot ions to cool, preventing their escape on the next cycle

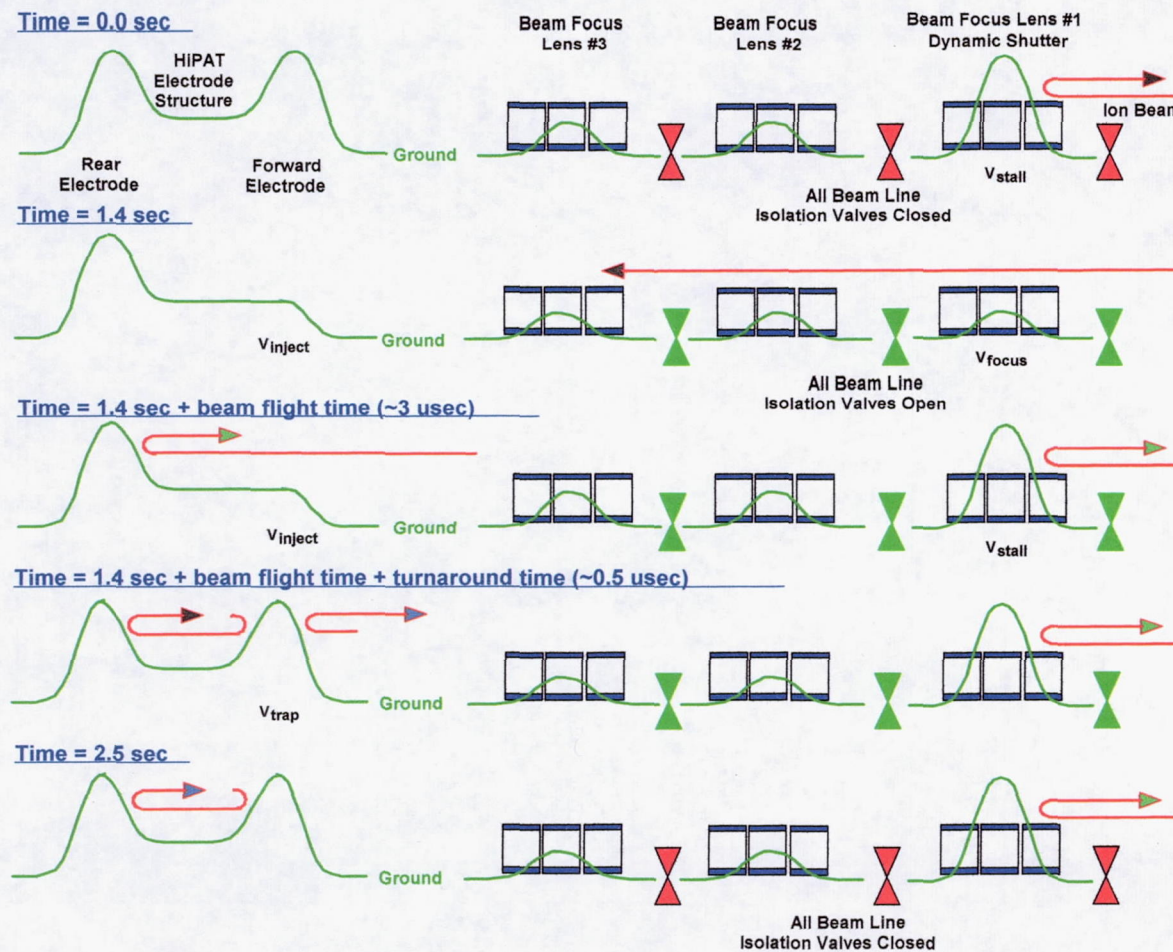




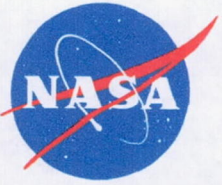


# Setup For Dynamic Capture

The HiPAT hardware uses the following dynamic system incorporating a series of valves, electrostatic lenses, and “trap door” electrodes.





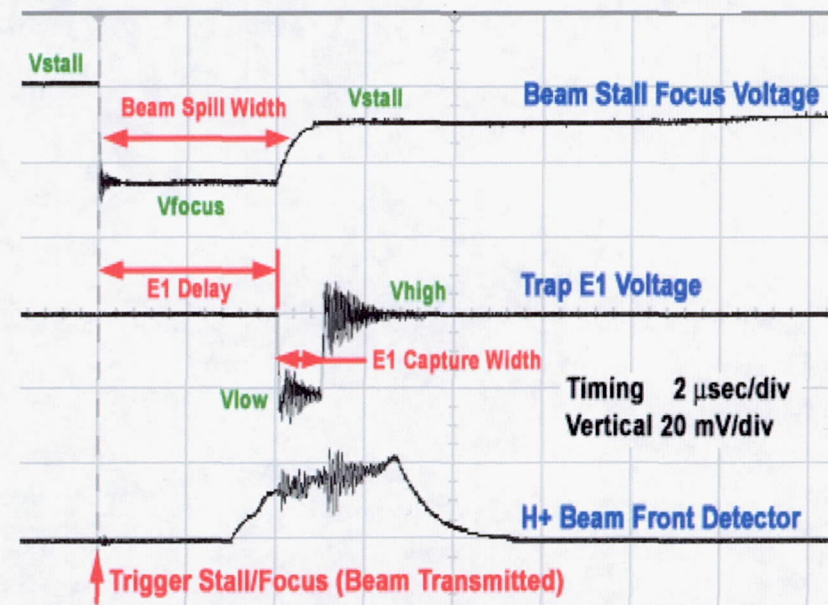


# Dynamic Capture of H<sup>+</sup> Ions

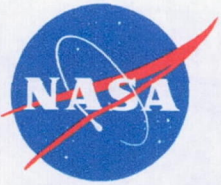
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The beam line connecting the ion sources to HiPAT has been configured for providing pulses of hydrogen ions. These pulses are captured by dynamically cycling the HiPAT trap.

- Beam line valves used to minimize gas loading...  $10^{-6}$  torr to  $10^{-11}$  torr (cycle time ~2.5 seconds)
- Focusing Einzel lens used as an electrostatic shutter. Triggering between Stall/Focus (cycle time as fast as 0.1 nanoseconds)
- Trap's forward electrode (E1) voltage collapses using dump timing circuit to capture a portion of the beam. (cycle time as fast as 0.1 microseconds)
- BNC 555 pulse timer used to synchronize timing of components. Behlke high speed HTS-301 transistor switches used.





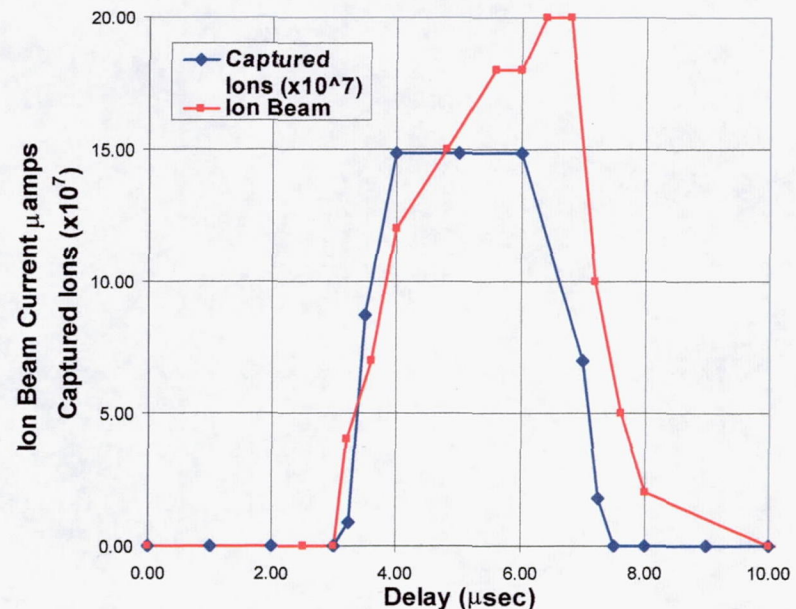


# Dynamic Capture

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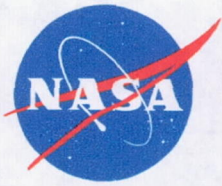
HiPAT dynamic capture system has successfully demonstrated confinement of hydrogen ions.

- Trap electrode (E1) cycle delayed varied with respect to initial ion transmission down beam line (stall/focus lens).
- Ion capture occurs only during interval where electrode cycling and the beam coincide.
- Results show  $\sim 1.5 \times 10^8$  ions captured during the center of the interval. Leading and trailing edges of ion beam sampling not sharp due to resistance/capacitance of pickup system.
- Data shows no appreciable ionization created by incoming ion beam (no ions extracted with small delay).
- Ionization of "Hot" captured beam while it cools still to be assessed.



- Beam spill width of 4 μsec, trap electrode cycle width of 1 μsec.
- Trap flat potential well 1kV (plasma column geometry) with end potentials at 3 kV.
- Ion beam set to ~2 kV energy with an intensity of ~20 μamps.



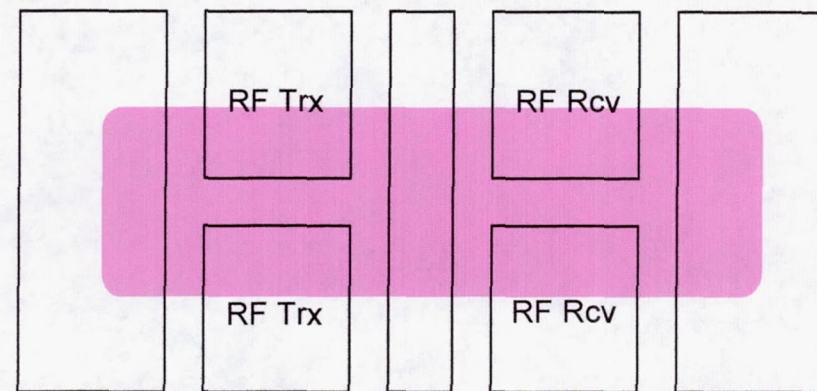
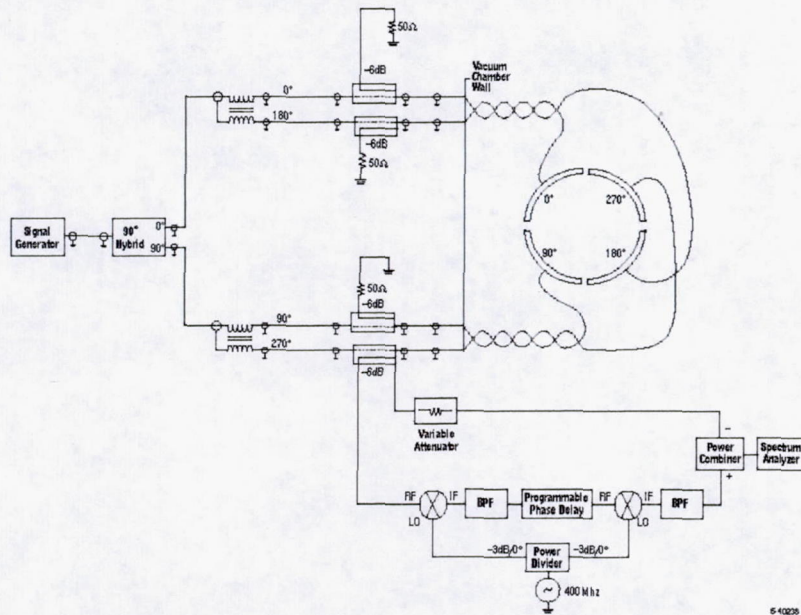


# Radio Frequency Particle Detection

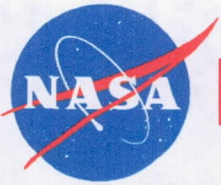
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Examine a non-destructive method for detection and diagnosis of trapped ions.

- Measure fundamental ion frequencies & amplitudes (function of containment fields).
- Apply radio frequency energy & examine the RF-to-Plasma interaction.
  - Two sets of sectioned electrodes serve as antenna for transmit and receive.
  - External low noise amplifiers, couplers, spectrum analyzer, and RF sweep generator.
  - Receiver average noise floor  $-130$  dBm with 10 kHz to 100 MHz bandwidth.
- Ultimate goal: Relate signal amplitude with quantity and species, use RF energy to stabilize ions increasing lifetime from minutes to weeks.
- Product: An autonomous computer driven ion health monitoring system for HiPAT.





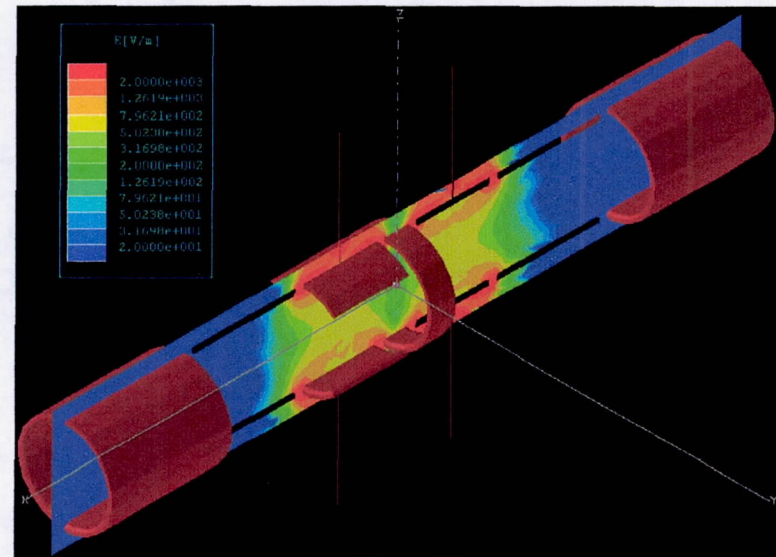
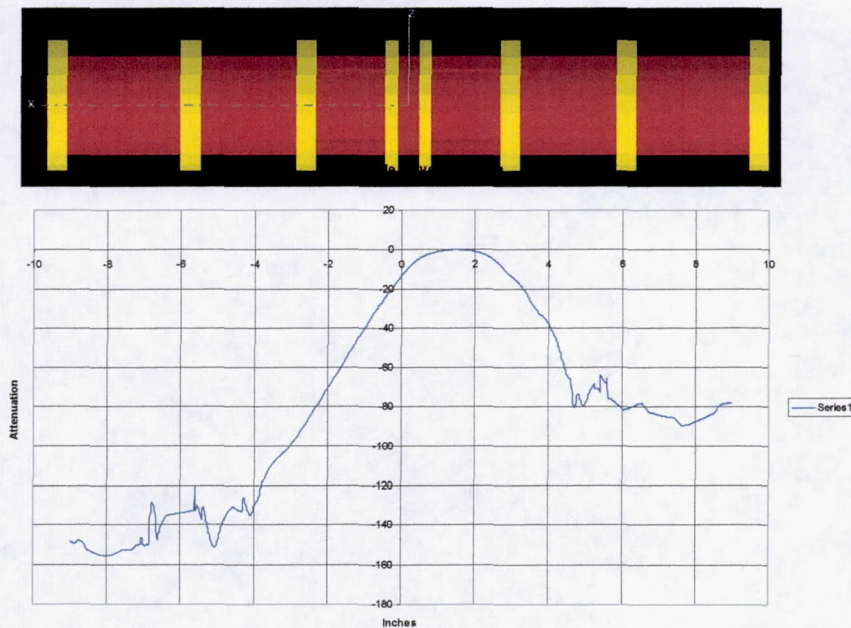


# Radio Frequency Antenna Modeling

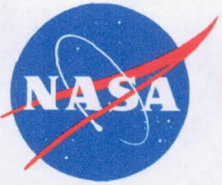
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## HiPAT electrode structure modeled

- Antenna characteristics modeled with EM circuit simulation package (Agilent HFSS)
- Preliminary topology shows that beyond 5 inches from the center the attenuation of the signal is approximately  $-80$  dB (normalized to maximum power coupling) at 10 MHz
- Eventually this simulation will include coupling to the plasma.





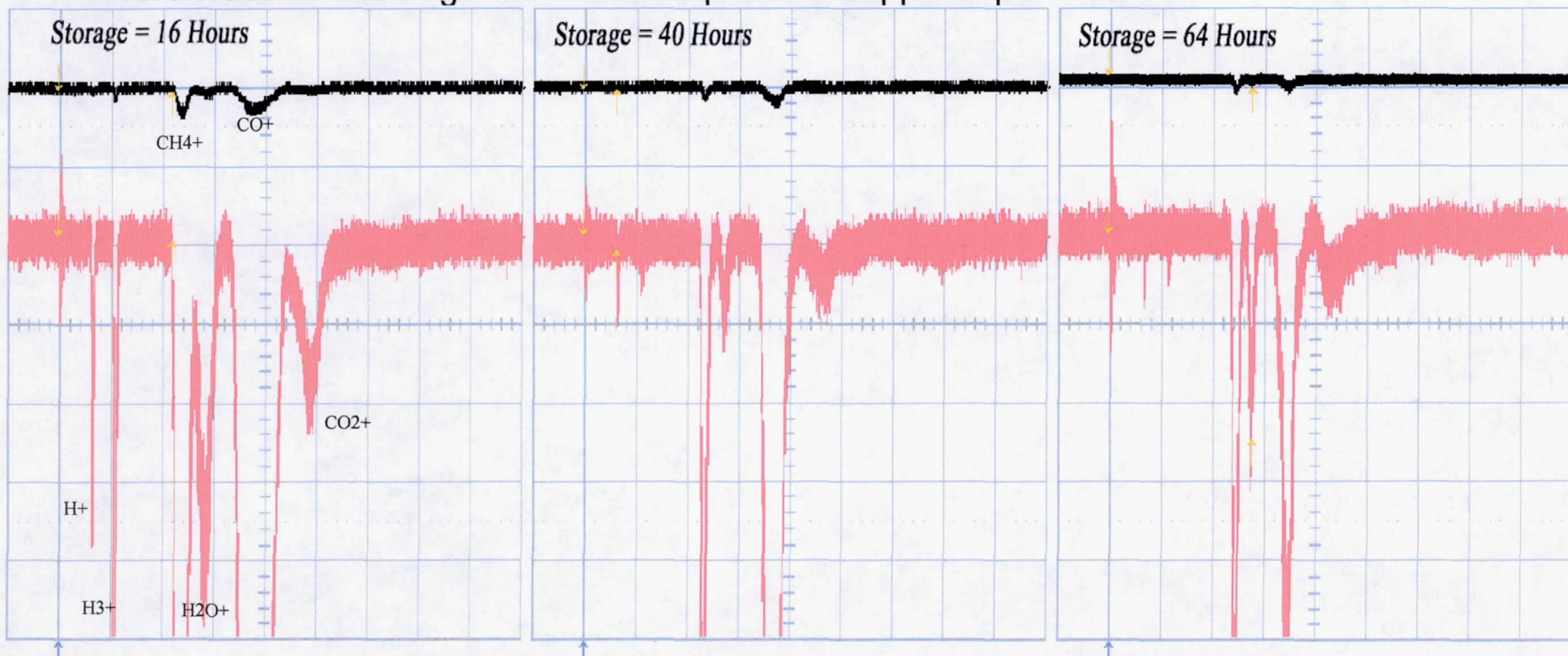


# R.F. Stabilization- Low Frequencies

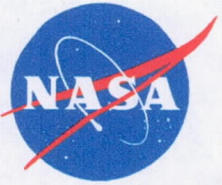
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Low frequency excitation was examined resulting in stabilization of trapped ions. Ranges of frequencies with varied amplitudes were investigated

- Low frequency excitation 50 to 250 kHz range (cloud rotation) appears to stabilize all species.
- Frequency ranges to stabilize specific ions (while excluding others) were not found: it was an “all or nothing” proposition.
- Baseline no RF tests – nearly all ions were gone within 16 hours.
- All tests used electron gun ionization to produce trappable particles.





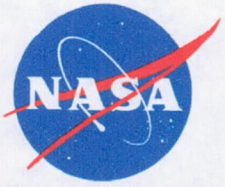


# Closing Remarks

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- The HiPAT system has been demonstrated to hold has been successfully demonstrated capture and containment of low numbers of ions.
- The NEC ion source system has provided a very nice mechanism of producing trappable ions (investigate increasing beam intensity in an effort to reduce stacking requirements).
- Focus to complete development of ion loading techniques ( $10^9$  to  $10^{11}$  range) with sufficient lifetime (order of minutes) to support research of the RF detection/stabilization system.
- Ongoing theoretical/experimental studies to identify plasma frequencies, densities and temperature with a goal of enabling predictable RF ion stabilization.





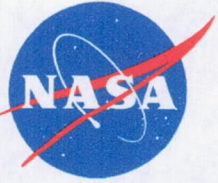
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# **BACKUP STUFF**

## **Propulsion Applications of Antimatter**

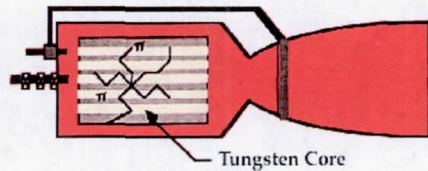




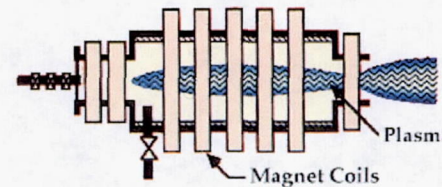
# Traditional Applications

## Conventional antimatter driven propulsion concepts

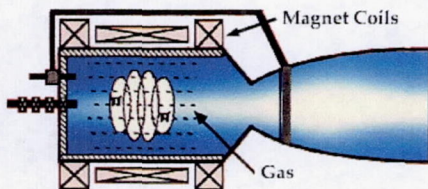
- These systems derive all their thrust from matter/antimatter annihilation
- Large amounts of antimatter would be required for operation (grams to metric tons)
  - $I_{sp}$  = Specific Impulse (propellant usage efficiency thrust/propellant weight flow rate)
  - $\eta_p$  = Efficiency of utilization (% of available annihilation energy)
- Solid Core: Limited by material temperature issues, dense heat exchanger high conversion eff
- Gas Core: Higher temperatures achieved, low gas density results in low conversion eff
- Plasma Core: Ionized gas with magnetic confinement, very low gas density lowest conversion eff.
- Beam Core: "Ultimate" system with no secondary fluids, magnets direct annihilation products directly



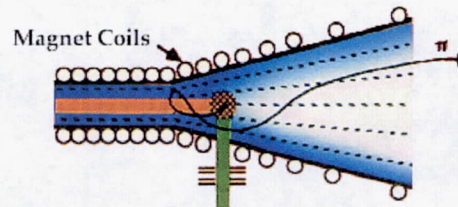
**Solid Core**  
 $I_{sp}$  - 1,000 sec  
 $\eta_p$  - 85%



**Plasma Core**  
 $I_{sp}$  -  $10^5$  sec  
 $\eta_p$  - 10%

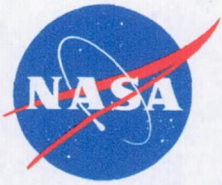


**Gas Core**  
 $I_{sp}$  - 2,000 sec  
 $\eta_p$  - 35%



**Beam Core**  
 $I_{sp}$  -  $10^7$  sec  
 $\eta_p$  - 60%

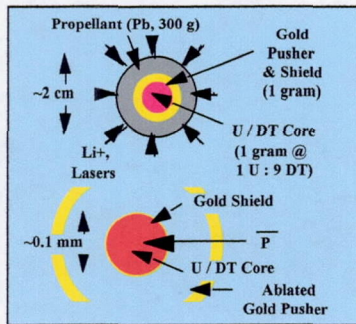




# Hybrid Applications

Hybrid antimatter systems are configured to derive most of their energy from fission and/or fusion reactions

- Acts as an “igniter” to initiating fission/fusion reactions lowering system driver mass requirements.
- Hybrid systems require less antimatter (1 to 100’s of  $\mu$ grams) than conventional approaches.



## *Antimatter-Catalyzed Micro-Fusion (ACMF)*

$I_{sp}$  - 13,500 sec (Specific Impulse)

$\eta_p$  - 15% (Propulsive energy utilization)

$\lambda$  - 0.7 (Vehicle structure/propellant mass ratio)

$\beta$  -  $1.6 \times 10^7$  (Fusion/annihilation energy ratio)

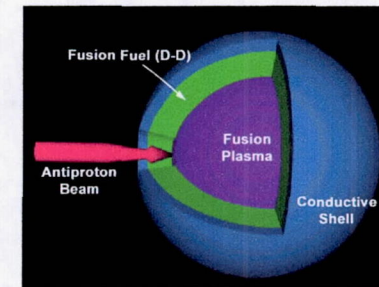
## *Antimatter-Magnetically Insulated Confined Fusion (AMICF)*

$I_{sp}$  - 200,000 sec (Specific Impulse)

$\eta_p$  - 10% (Propulsive energy utilization)

$\lambda$  - 2.3 (Vehicle structure/propellant mass ratio)

$\beta$  -  $5.0 \times 10^3$  (Fusion/annihilation energy ratio)



## *Antimatter-Initiated Micro-fusion (AIM)*

$I_{sp}$  - 67,000 sec

$\eta_p$  - 84%

$\lambda$  - 0.2

$\beta$  -  $10^5$

D-He3

$I_{sp}$  - 61,000 sec

$\eta_p$  - 69%

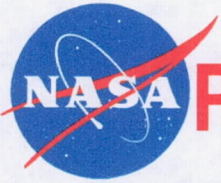
$\lambda$  - 0.3

$\beta$  -  $2.2 \times 10^4$

D-T







# PROPULSION APPLICATION (Cont.)

MARSHALL SPACE  
FLIGHT CENTER  
HUNTSVILLE, AL

## Comparative performance of antimatter based propulsion concepts

