EXTINCTION CRITERIA FOR OPPOSED-FLOW FLAME SPREAD IN A MICROGRAVITY ENVIRONMENT

SUBRATA BHATTACHARJEE CHRIS PAOLINI San Diego State University, San Diego, California

and

KAZUNORI WAKAI SHUHEI TAKAHASHI Gifu University, Gifu, Japan

INTRODUCTION: A simplified analysis is presented to extend a previous work [1] on flame extinction in a quiescent microgravity environment to a more likely situation of a mild opposing flow. The energy balance equation, that includes surface re-radiation, is solved to yield a closed form spread rate expression in terms of its thermal limit, and a radiation number that can be evaluated from the known parameters of the problem. Based on this spread rate expression, extinction criterions for a flame over solid fuels, both thin and thick, have been developed that are qualitatively verified with experiments conducted at the MGLAB [2] in Japan. Flammability maps with oxygen level, opposing flow velocity and fuel thickness as independent variables are extracted from the theory that explains the well-established trends in the existing experimental data [3].

Thermal Regime: An energy balance for the solid phase control volume of Fig. 1 can be written as.

$$\lambda_g \frac{\left(T_f - T_v\right)}{L_g} L_g - \varepsilon \sigma \left(T_v^4 - T_\infty^4\right) L_g \sim V_f \rho_s c_s \tau_h \left(T_v - T_\infty\right)$$
(1)

where, T_f and T_v are characteristic flame and vaporization temperature, τ_h is the thickness of the heated layer, and $L_g = \alpha_g / (V_f + V_g)$ is the gas-phase length scale. For thin fuels in the thermal limit, $\tau_h = \tau$ and $\varepsilon = 0$ produces the de Ris solution $V_{f,\text{th,thn}} \sim (\lambda_g / \rho_s c_s \tau) F$, where $F \equiv (T_f - T_v) / (T_v - T_\infty)$. Using $V_{f,\text{th,thn}}$ to non-dimensionalize V_f , $\eta_f \equiv V_f / V_{f,\text{th,thn}}$ Eq. (1) can be expressed in non-dimensional form as follows.

$$\left(\eta_{f}^{2} + \eta_{f}\eta_{g}\right)\frac{\tau_{h}}{\tau} - \left(\eta_{f} + \eta_{g}\right) + \Re_{0} \sim 0 \text{ ; where, } \Re_{0} \equiv \frac{1}{F^{2}}\frac{\rho_{s}c_{s}}{\rho_{g}c_{g}}\frac{\varepsilon\sigma\tau}{\lambda_{g}}\left(\frac{T_{v}^{4} - T_{z}^{4}}{T_{v} - T_{z}}\right), \eta_{g} \equiv \frac{V_{g}}{V_{f,\text{th,thn}}}$$
(2)

The thermal thin limit $\eta_{f,\text{th,thin}} \sim 1$ is recovered when $\Re_0 = 0$ and $\tau_h = \tau$. To obtain a more general solution τ_h / τ for a thick fuel can be scaled as

$$\frac{\tau_h}{\tau} \sim \frac{\sqrt{\alpha_s t_{\text{res},s}}}{\tau} \sim \sqrt{\frac{\alpha_s L_g}{\tau^2 V_f}} \sim \frac{\Omega}{F} \frac{1}{\sqrt{\eta_f (\eta_f + \eta_g)}} \quad \text{where, } \Omega \equiv \sqrt{\frac{\lambda_s \rho_s c_s}{\lambda_g \rho_g c_g}} \quad .$$
(3)

Substituting this into Eq. (2) and still ignoring radiation, we obtain the thermal limit for semi-infinite fuel beds.

$$\eta_{f,\text{th,thick}} \sim \frac{F^2}{\Omega^2} \eta_g \left(1 - \frac{F^2}{\Omega^2} \right)^{-1} \sim \frac{F^2}{\Omega^2} \eta_g \quad \text{if } \tau \ge \tau_h; \text{ or, } \tau \ge \frac{\lambda_s}{\rho_g c_g V_g F}; \text{ or, } \eta_g \ge \frac{\Omega^2}{F^2} \tag{4}$$

The simplification above is achieved because for both PMMA and cellulose it can be shown that $F < \Omega$. Equation (4) also provides a criterion for transition between the thin and the thick limit for $\eta_g \gg 1$. Prediction from Eq. (4)

is plotted in Fig. 2 showing the transition from the thin to the thick limit in the thermal regime. Not much data in the thick-thin transitional region is available to verify this simple transition criterion.

Radiative Regime: The energy balance equation, Eq. (2), is solved in both the thick and thin limit producing

Thin Limit:
$$\eta_{f,\text{thin}} \sim \frac{1-\eta_g}{2} + \frac{1}{2}\sqrt{\left(1+\eta_g\right)^2 - 4\Re_0}$$
; Thick Limit: $\eta_{f,\text{thick}} \sim \frac{F^2}{\Omega^2}\eta_g \left(1-\frac{\Re_0}{\eta_g}\right)^2$; (5)

These results are plotted in Figs. 3 and 4 for several values of the radiation parameter \Re_0 . A number of important features of the radiative effects on flame spread rates are revealed by these plots. When $\Re_0 > 0$, the slope of the spread rate curves decreases with opposing velocity for thin fuels while this trend is completely reversed for thick fuels. The MGLAB data [2] for flame spread over thin PMMA, shown in Fig 5, support this predicted trend for thin fuels. The DARTFIRE experiments [4] for flame spread over thick PMMA lends supports to the trends predicted by Fig. 4.

Obvioulsy, for $\mathfrak{R}_0 = 0$ and/or $\eta_g \to \infty$, the thermal limits are recovered with $\eta_{f,\text{thin}} = 1$ and $\eta_{f,\text{thick}}$ being proportional to η_g . To establish a criterion for the transition between the thermal and radiative regimes, we simplify Eq. (5) assuming $\eta_g \gg 1$. If the spread rate is non-dimensionalized by the corresponding thermal limit, Eq. **Error! Reference source not found.** for both the thick and thin limit can be shown to approach the same form.

Thin and Thick Fuels: For
$$\eta_g \gg 1$$
, $\eta'_f \equiv \frac{V_f}{V_{f,\text{thermal}}} \sim \left(1 - \frac{\Re_0}{\eta_g}\right)^2 \sim 1 - 2\Re_V$; where, $\Re_V \equiv \frac{\Re_0}{\eta_g}$ (6)

A single parameter \mathfrak{R}_{V} , therefore, controls the radiative effects on the spread rate for both thermally thin and thick fuels. η'_{f} from Eq. (6) is plotted in Fig. 6 against versus $1/\mathfrak{R}_{V}$, so that the abscissa is proportional to V_{g} . Superposed on this figure are experimental spread rates from MGLAB experiments, only part of which were previously reported [2]. Although the spread of the data around the prediction of Eq. (6) is substantial, the onset of radiative effects seems to be well correlated by the analytical prediction.

Extinction Criteria: The spread rate expressions of Eq. (5) can be used to establish criterion for flame extinguishment. As can be seen from Figs. 3 and 4, there are two types of extinction behavior. For $\eta_g \ge 1$, in both the thin and thick limit, steady flame cannot be sustained provided $\eta_g > \Re_0$, a criterion that is independent of fuel thickness. For $\eta_g < 1$, the thick fuel criterion remains unaltered. However for thin fuels, the spread rate assumes complex values, an indication of extinguishment, when $\eta_g < 2\sqrt{\Re_0} - 1$. For flame spread over PMMA, these criteria are combined in the flammability map of Fig. 7. Note that for a critical thickness can be calculated from the relation $\eta_g = \Re_0 = 1$, beyond which extinction is independent of fuel thickness, thereby, defining a radiatively thick fuel.

Conclusion

In this article we present a simplified analysis to develop for the first time a closed-form expression for the spread rate and extinction criterion for flame spread over condensed fuels in a mild opposing-flow microgravity environment. The results presented are supported by experiments on thin PMMA conducted in the MGLAB.

Acknowledgement: Support from NASA, Glenn Research Center, with Dr. Sandra Olson as the contract monitor, is gratefully acknowledged.

REFERENCES: [1] Bhattacharjee, S., Takahashi, Wakai, K., A., *Prediction of a Critical Fuel Thickness for Flame Extinction in a Quiescent Microgravity Environment,* Combustion and Flame, to appear, 2003

[2] Takahashi, S., Kondou, M., Wakai, K., A., Bhattacharjee, S., Proceedings of the Combustion Institute, Vol 29, 2002.)

[3] Olson, S.L., Ferkul, P.V., and T'ien, J.S., Proceedings of the Combustion Institute, 22:1213, (1988)

[4] Altenkirch, R.A, Olson, S., and, Bhattacharjee, S., "Diffusive and Radiative Transport in Fires", NASA Contract NCC3-221 (1993-1998)



Fig. 1 Control volumes at the flame leading edge in the gas and the solid phases.



Fig. 2. Non-dimensional spread rate in the thermal regime.



Fig. 3. Spread rate as a function of η_g and \Re_0 as predicted by Eq. (7). Opposed-flow flame spread extends down to $\eta_g = -1$.



Fig. 4. Spread rate for thick fuel as a function of η_g and \Re_0 as predicted by Eq. (7). The spread rate is zero (extinction) for $\eta_g < \Re_0$.



Fig. 5. Non-dimensional experimental spread rate [9] as a function of η_g for different oxygen mole fractions and fuel half-thickness. Note that in this plot the highest experimental spread rate is used to normalize V_f and V_g instead of the theoretical thermal limit.



Fig. 6. Prediction of the non-dimensional spread rate η_f plotted as a function of

inverse of $\mathfrak{R}_{0,thin}$ from Eq. (10). The prediction is compared with the spread rate data from the MGLAB experiments.