

- **Chuck Weisbin**

Multi-Mission Strategic Technology Prioritization Study

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"Systematic Technology Prioritization For New Space Missions"

Humphrey's Half Moon Inn, San Diego, CA

Jet Propulsion Laboratory
California Institute of Technology
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- **S. Prusha for assisting in selection of ECS technologies to analyze; M. Feather for providing information about correlations of tasks and needs**

Study Staff & Roles

➤ JPL

- J. Derleth, Mission & Technology Portfolio Optimization
- A. Elfes, ECS Data & Analysis
- B. Kennedy, ECT Data & Analysis
- R. Manvi, Tech Life Cycle & Risk Management Model
- K. Shelton, Mission & Technology Data Base
- J. H. Smith, Integrated Risk Analysis
- G. Rodriguez, System Analysis

➤ GSFC staff (M. Steiner, J. Azzolini, J. Mapar, C. Stromgren)

Study Objectives

- **Perform a pilot study of sufficient breadth which demonstrates in an auditable fashion how advanced space technology development can best impact future NASA missions**
 - Include wide spectrum of missions & technologies
 - Can add new missions & technologies easily
 - Optimize technology portfolios
 - Lead to rapidly prototyped example
- **Show an approach to deal effectively with inter-program analysis trades**
- **Explore the limits of these approaches and tools in terms of what can be realistically achieved (scope, detail, schedule, etc.)**

Technology Portfolio Optimization Approach

- **Collect performance data for many individual technologies; each data input is viewed as a statistical sample representing an expert assessment**
- **Group the technological data into a tree-like hierarchical model to predict “integrated” system, mission, and multi-mission impact of individual technologies**
- **Search computationally for technology portfolios with optimal science return, risk and cost impact**
- **Investigate sensitivity of the optimal portfolio to changes in available budget levels**

Major Study Challenges

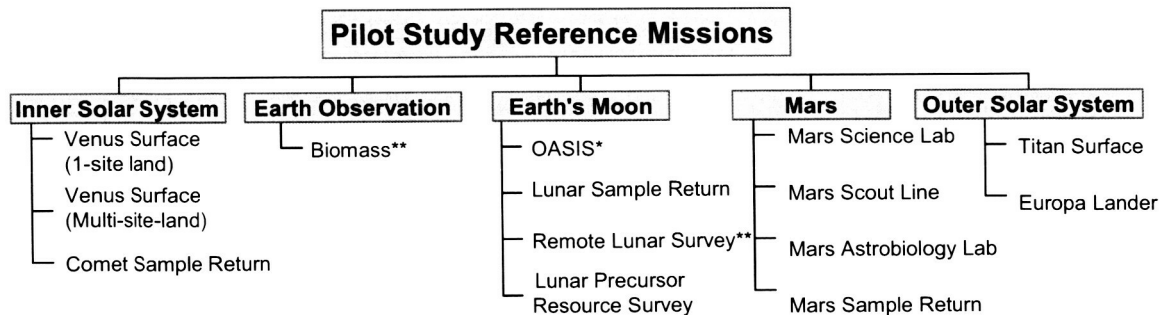
- **Reference Missions: assess mission value; characterize capability requirements**
- **Technology Projections: characterize performance; manage widely dispersed and non-uniform data**
- **Uncertainty: incorporate & manage widespread uncertainty**
- **ROI Measures: formulate suitable value function for portfolio analysis**
- **Layers of Abstraction: choose and maintain appropriate level of analytical abstraction**
- **Technological Boundaries: boundaries of technology domains not clearly marked**
- **Many Scales: large differences in cost and performance scales for different technologies**
- **Performance Parameters: not fully understood for some technologies**
-

Implementation Approach

- **Iterative in three phases (keep eye on big picture early, and continuously)**
 - Phase 1 minimalist multi-mission set; ECT/ECS technologies
 - Phase 2 more extensive set of missions & technologies (June 04)
 - Phase 3 completion of full study (December 04)
- **Maintain high degree of connectivity**
 - Space Architect
 - Revolutionary Mission Concepts
 - Advanced Space Technology Programs
 - Enterprises
 - Centers
 - Etc.

Pilot Study Reference Missions

(Organized by Science-Site Location)



- Initial reference mission set as of April 15, 2004
- More missions and enabling technologies will be added throughout the period of performance of the study

* OASIS is a near Earth transportation infrastructure that enables access to the Moon. It consists of: a Hybrid Propellant Module, a Chemical Propulsion Module, a Solar Electric Propulsion Module, and a Crew Transport Vehicle.

** GSFC contribution to this study focuses on these missions

Reference Missions & Major Challenges

(Minimalist Mission Set for PHASE I)

Reference Mission Classes (not listed in order of priority)	Major Challenges
Earth's Moon: Orbital Aggregation and Space Infrastructure Systems (OASIS); Lunar Remote Survey; Lunar Surface Missions; etc.	Deep Space Robotic Rendezvous & Docking; Long Term Cryogenic Fuel Storage in Space (>2 years); Long Life Ion Engines(>15 K-hours)
Mars Surface: (e.g. Mars Science Laboratory; Astrobiology Field Lab; Mars Sample Return; etc.)	Long-Range, Long-Life Mobility (10's of kilometers, >600 sols); Substantive Sample Collection and Return (>1kg, 0<depth<100m subsurface)
Earth Observation: Biomass	Lidar/Radar Instrument Systems; Multi-Spectral Scanner; Sensor Webs & Data Fusion
Outer Solar System: Titan Surface; Europa Lander	Extreme Environments; Sub-Surface Ice Mobility
Inner Solar System: Venus surface; comet sample return	Extreme Environments (460C temp; 90 bar pressure; sulfuric acid clouds at 50 km)

➤ **Technologies to be evaluated will include:**

- Technological products in several discipline fields (aimed at operational flight system implementation (e.g. advanced materials, structures, etc.))
- Risk assessment tools and infrastructure to allow for risk quantification, and risk mitigation during an entire mission life-cycle, but that do not necessarily appear in the flight system implementation (e.g. risk management methods)

Enabling Technologies for Which Data Has Been Collected to Date

- **Extreme Temp & Pressure Components, Thermal Control, Pressure-Vessel-Encapsulated Electronics (Venus)**
- **Electric & Chemical Propulsion; Reaction Control; Multifunction Structures; Fuel Storage & Control; Syntactic Foams, Formation Flying (OASIS)**
- **Entry Descent & Landing; Surface, Aerial, Subsurface Mobility; Manipulation, Drilling, Sampling (Mars, Titan, Comet, Lunar Surface)**
- **In-Space Inspection, Maintenance, Assembly (OASIS, Large Observatory Platform, Gateway, Space Solar Power)**
- **Risk Methods, Tools and Workstation; Mishap Anomaly Data Base; Complex Systems Research; Risk Characterization & Visualization; etc. (All Reference Missions)**

Enabling Technology Areas

(for which data has been collected to date)

Enabling Technology Areas	Missions
Electric & Chemical Propulsion; Reaction Control; Multifunction Structures; Fuel Storage & Control; Syntactic Foams, Formation Flying; In-Space Robotic Inspection, Maintenance, Assembly	OASIS
Entry Descent & Landing; Surface, Aerial, Subsurface Mobility; Manipulation, Drilling, Sampling	Mars, Earth's Moon, Titan, Comet
Risk Methods, Tools & Workstation; Mishap Anomaly Data Base; Complex Systems Research; Risk Characterization & Visualization; etc.	All
Extreme Temp & Pressure Components, Thermal Control, Pressure-Vessel-Encapsulated Electronics	Venus, Titan, Europa

Technology Areas are Decomposed into Many Sub-Areas & Performance Parameters

A Few Typical Technology Areas	A Few Typical Technology Sub-Areas	A Few Typical Performance Parameters
Multi-Function Structures	Modular, Distributed Structures, Deployable Structures, etc.	Contract/Extend (cm), Power per Mass (W/kg), etc.
Fuel Storage & Control	On Orbit Cryogenic Fuel Transfer, Tank Pressure Control, Fuel Storage, etc.	Flow Rate (kg/min), Pressure (kPa), Time (yrs), etc.
Subsurface Ice Mobility	Range, Radiation Dose, Payload Capacity, Ambient Pressure, etc.	Distance (km, mRads), Mass (kg), Pressure (atm), etc.
Extreme Temperature & Pressure Components	High Temperature Electronics, Permanent Magnets, Energy Storage, etc.	Temperature (Celsius), Pressure (Bars), Energy Density (Whr/l) etc.
Risk Methods, Tools & Workstation	Model Based Risk Analysis, Mission Risk Profiling Capability, etc.	Accessibility, applicability to multiple mission phases, risk mitigation coverage

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Mission & Technology Data Base

Mission Parameters				Years Start to Mission I										Years Start to Mission II						
level	metric	unit	partnr	valr	TRL	need	mean	worst	best	TRL	Yes	SM	need	mean	worst	best	TRL	Yes	SM	
0	# Yrs Survival	#	0.5	3																
0	# Landing Sites	#	1	3																
0	# Samples Per Site	#	1	3																
0	Projected # of Years to Phase A	Years	N/A	N/A	8	8	10	5	N/A	N/A	N/A	N/A	15	15	20	10	N/A	N/A	N/A	N/A
Technology				Mission One										Mission Two, Etc.						
1	Extreme Temp & Pressure Components (ASIC) Others																			
2	Sensors Operating at High Temp/Pressure																			
3	Temperature Sensors																			
4	Operating Temperature	degree Celsius		160	3	160	180	160	500			5	125	160	180	160	500	6	5	1
4	Operating Pressure	bar		1	3	90	90	80	100			5	125	90	120	80	150	6	5	1
3	Pressure Sensors																			
4	Operating Temperature	degree Celsius		160	3	160	180	160	500			5	125	160	180	160	500	6	5	1
4	Operating Pressure	bar		1	3	90	90	80	100			5	125	90	120	80	150	6	5	1
3	Position Sensors																			
4	Operating Temperature	degree Celsius		160	3	160	180	160	500			5	125	160	180	160	500	6	5	1
4	Operating Pressure	bar		1	3	90	90	80	100			5	125	90	120	80	150	6	5	1
5	Operating Temperature	degree Celsius		150	3	160	160	150	470	6	5	125	160	160	150	470	6	5	1	1
5	Operating Pressure	bar		90	3	90	90	80	100			5	125	90	90	80	100	6	5	1
5	Operating Temperature	degree Celsius		160	3	160	160	150	470	6	5	125	160	160	150	470	6	5	1	1
5	Operating Pressure	bar		90	3	90	90	80	100			5	125	90	90	80	100	6	5	1
5	Operating Temperature	degree Celsius		160	3	160	160	150	470	6	5	125	160	160	150	470	6	5	1	1
5	Operating Pressure	bar		90	3	90	90	80	100			5	125	90	90	80	100	6	5	1
5	# Sensors Integrated	#				4	4	3	5	6	5	2								
5	Operating Temperature	degree Celsius		500	3	500	500	180	510	6	5	1	500	500	180	510	6	5	1	1
5	Operating Pressure	bar		100	3	100	100	80	100			5	100	100	80	100	6	5	1	1
5	Operating Temperature	degree Celsius		160	3	160	160	150	470	6	5	1	160	160	150	470	6	5	1	1
5	Operating Pressure	bar		90	3	90	90	80	100			5	90	90	80	100	6	5	1	1
5	Max Energy Product	W/kg		26	3	26	26	18	12	6	5	1	26	26	18	12	6	5	1	1
5	Conductivity	W/mK		10000	3	10000	10000	8000	12000	6	5	1	10000	10000	8000	12000	6	5	1	1
5	Max Operating Temperature	degree Celsius		160	3	160	160	150	470	6	5	1	160	160	150	470	6	5	1	1
5	Energy Density	Wh/kg		200	3	200	200	150	250	6	5	2	200	200	150	250	6	5	2	2
5	Operating Temperature	degree Celsius		160	3	160	160	150	470	6	5	1	160	160	150	470	6	5	1	1
5	Shell Lifetime	Yrs		5	3	5	5	4	6	6	5	1	5	5	4	6	6	5	1	1
5	# of Recharge Cycles	#		100	3	100	100	80	120	6	5	1	100	100	80	120	6	5	1	1
5	Energy Density	Wh/kg		200	3	200	200	180	220	6	5	1	200	200	180	220	6	5	1	1
5	Operating Temperature	degree Celsius		160	3	160	160	150	470	6	5	1	160	160	150	470	6	5	1	1
5	Shell Lifetime	Yrs		1	3	5	5	4	6	6	5	1	5	5	4	6	6	5	1	1
5	# of Recharge Cycles	#		100	3	100	100	80	120	6	5	1	100	100	80	120	6	5	1	1

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Mission & Technology Data Base

-- Current Size Summary --

- **Size of Mission & Technology Capability Data Base (as of April 15, 2004)**
 - 13 missions covering wide spectrum of NASA strategic plans
 - 23 technology areas (structures, energetics, extreme environments, surface mobility, etc.)
 - 86 technology sub-areas (batteries, payload capacity, thermal control, etc.)
 - 167 technological performance parameters (power density, operating temperature, etc.)
- **Remarks About Data Base**
 - Current data set is more detailed in some areas than in others
 - More technologies & detail will be collected in subsequent phases
 - Our analysis methods can handle data sets with non-uniform detail

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Risk Related Requirements

(from Point of View of a Project Manager)

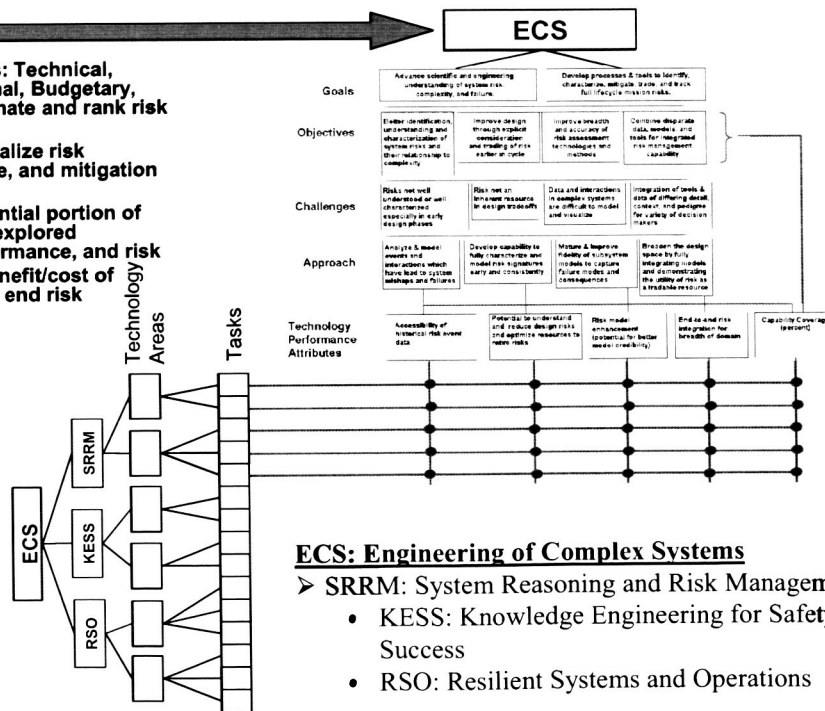
• Risk Management Must:

- Delineate major risks: Technical, Human, Organizational, Budgetary, and Schedules ;estimate and rank risk levels
- Provide ways to visualize risk elements, time profile, and mitigation strategies
- Assure that the systems and trade analysis includes cost, performance, and risk
- Provide auditable benefit/cost of implementing begin-to-end risk mitigation strategies

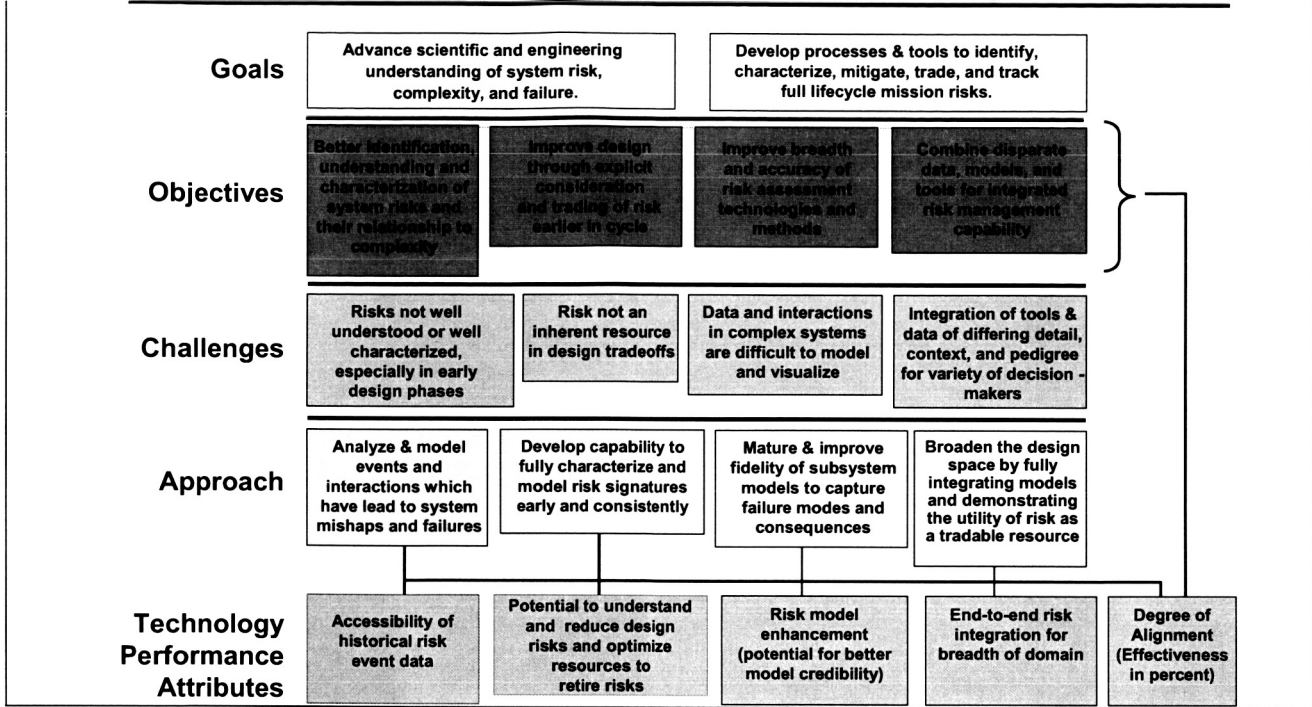
Connecting Risk Technologies to Requirements

Requirements:


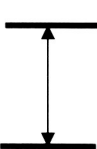


- Delineate major risks: Technical, Human, Organizational, Budgetary, and Schedules; estimate and rank risk levels
- Provide ways to visualize risk elements, time profile, and mitigation strategies
- Assure that a substantial portion of the design space is explored including cost, performance, and risk
- Provide auditable benefit/cost of implementing end to end risk mitigation strategies



System Reasoning and Risk Management (SRRM) Project Executive Summary



Attribute Definitions

Accessibility of risk data	Best Case  Worst Case	10 5 0	Easy to use DB spans multiple mission/projects with risk events categorized for search. DB may be limited to specific category or series of missions. Supporting data/verifications are anecdotal (narrative) format without categories of risk events for easy search. May require further processing to another format.
Potential to reduce design risks	Best Case  Worst Case	10 5 0	Technology helps to identify and reduce risks during early phases of project (Phase A/B) with potential to dramatically reduce overall project costs by reducing rework. Technology helps identify/reduce mission risks for Phase C/D; Large potential cost benefits if used. Provides a screen that limits potential risks from passing CDR. Technology helps identify technology development or subsystem risks, but may or may not influence overall system risk.
Risk model enhancement	Best Case  Worst Case	10 5 0	Technology provides new approach for addressing design risk life-cycle or part of life-cycle not previously addressed (e.g., mgmt, org. risks) Technology either provides new, more effective approach for risk analysis or fills missing gap in temporal or breadth of risk analyses (but not both) Technology does not address missing gap in design life-cycle.
End-to-end risk integration	Best Case  Worst Case	10 5 0	Technology provides synergistic integration with other tools and databases fully compatible with emerging design environments (temporal and breadth). Risk technology allows interaction with common databases but cannot be integrated with other stand-alone applications. Technology is stand-alone; focused, narrow; little breadth or temporal range, databases are separated with little or no connectivity. Integration difficult.

All SRRM Technology Areas Are Included for the Pilot Study

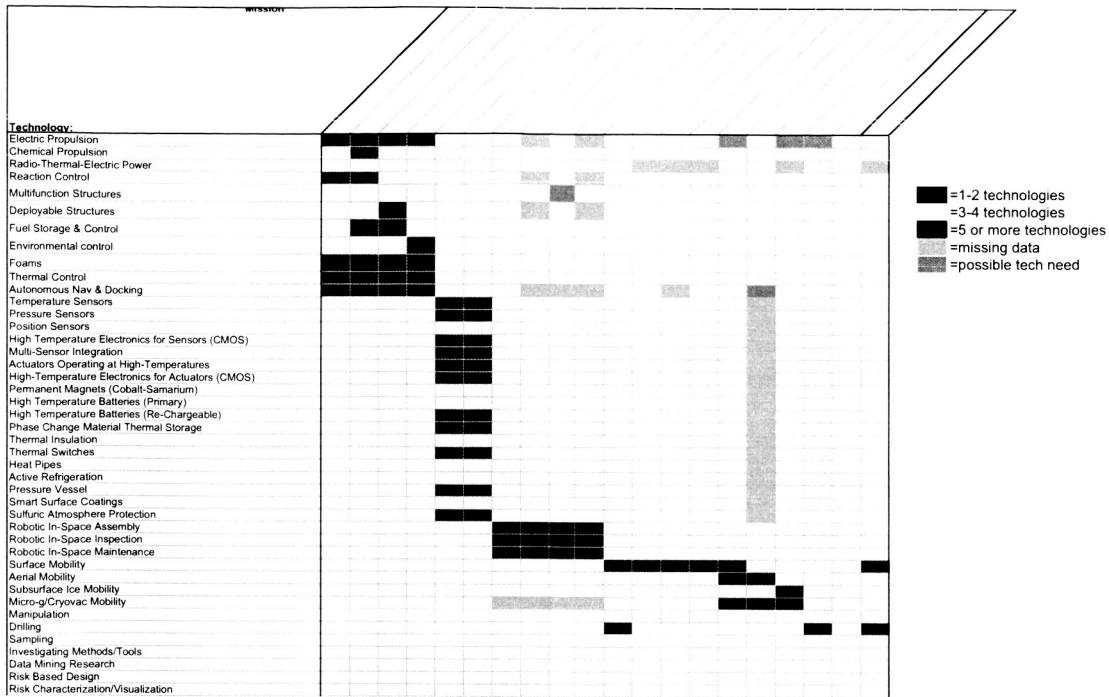
1. Risk Methods/Tools (RMT)
2. Risk Workstation (RWS)
3. Mishap/Anomaly Database (MAIS)
4. Model-Based Hazard Analysis (MBHA)
5. System Complex Research (SCR)
6. Risk Characterization/Visualization (RCV)
7. Risk-Based Design (RBDO)
8. Data Mining Research (DMR)
9. Investigation Methods/Tools (IMT)

Typical SRRM Technology Area Data*

Technology	Level	Metric	Unit	Polarity	SOA	Low	ML	High	\$M
		How performance is measured	What unit performance is measured in	+ = Better if performance is higher - = Better if performance is lower	Current state-of-the-art for similar technologies	Technologist's estimate of low, most likely, and high values of what will be provided to the mission			How much the technologist needs to achieve TRL 6 in \$M
ECS	1								
SRRM	2								
RISK Methods & Tools	4	Accessibility of Historical Risk Event Data	0-10	+	4	7	8	9	2
		Potential to Understand and Reduce Design Risks and Optimize Resources to Retire Risk	0-10	+	1	7	8	9	
		Risk Model Enhancement (Potential for Better Model Credibility)	0-10	+	2	9	10	10	
		End-to-end Risk Integration for Breadth of Domain	0-10	+	2	8	9	10	
		Extent of Needs Covered	0-1	+	0.5	0.7	0.8	0.9	

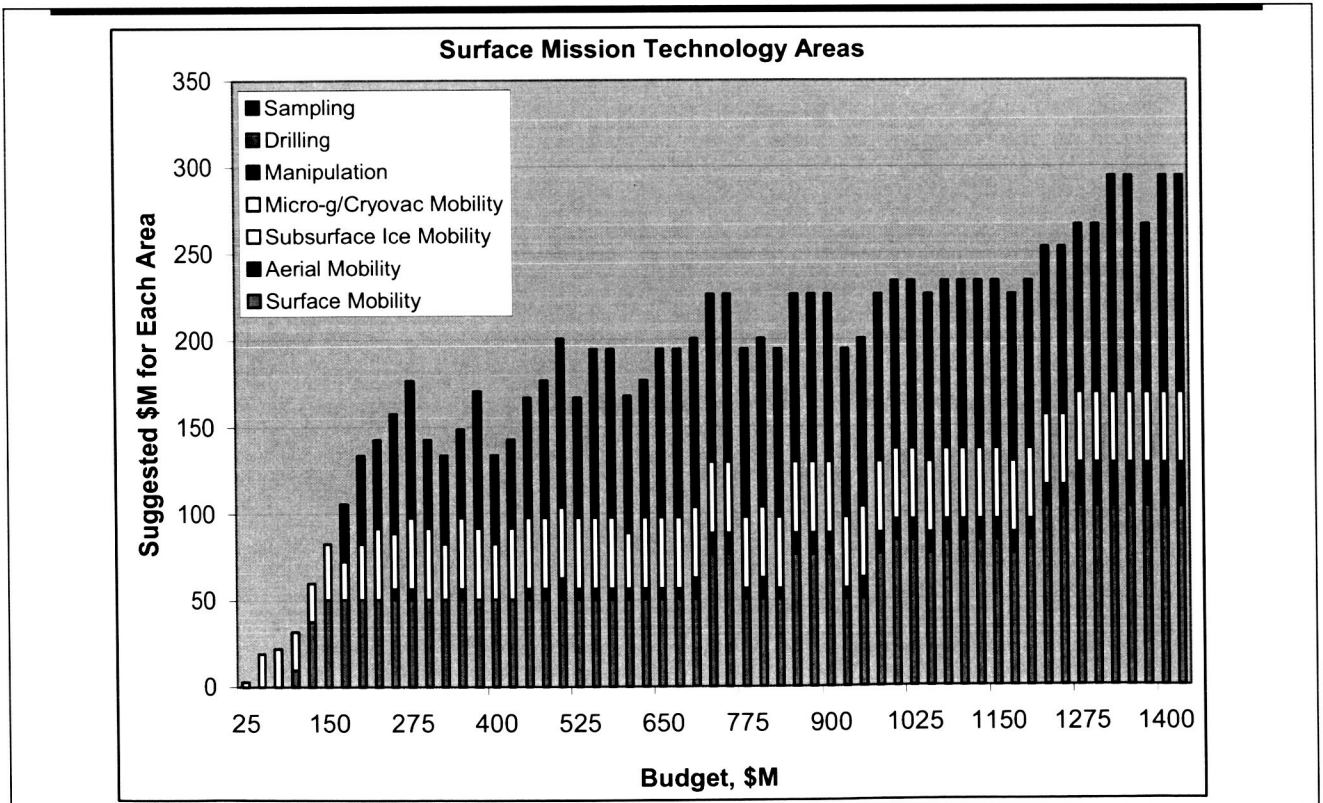
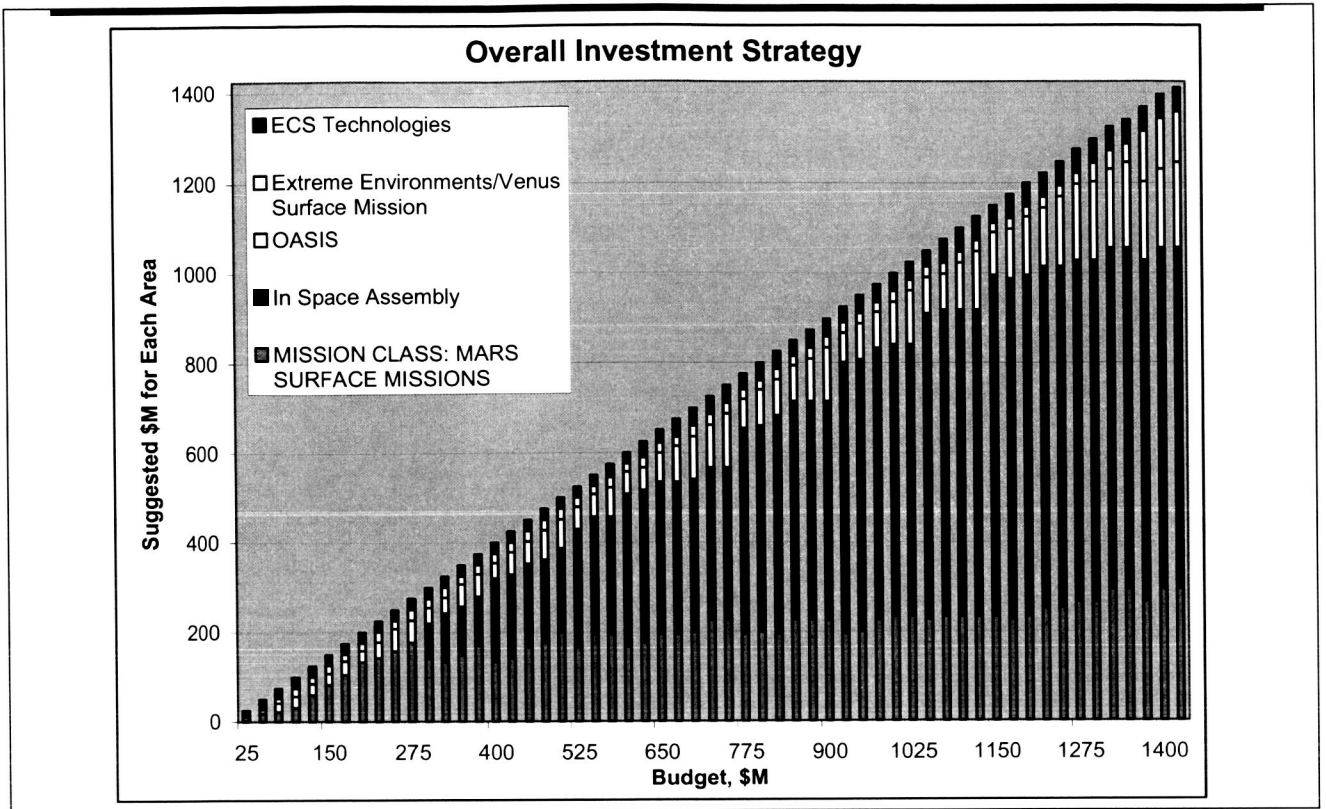
*SRRM data cast in same format used for all other technologies (shown in slide 14)

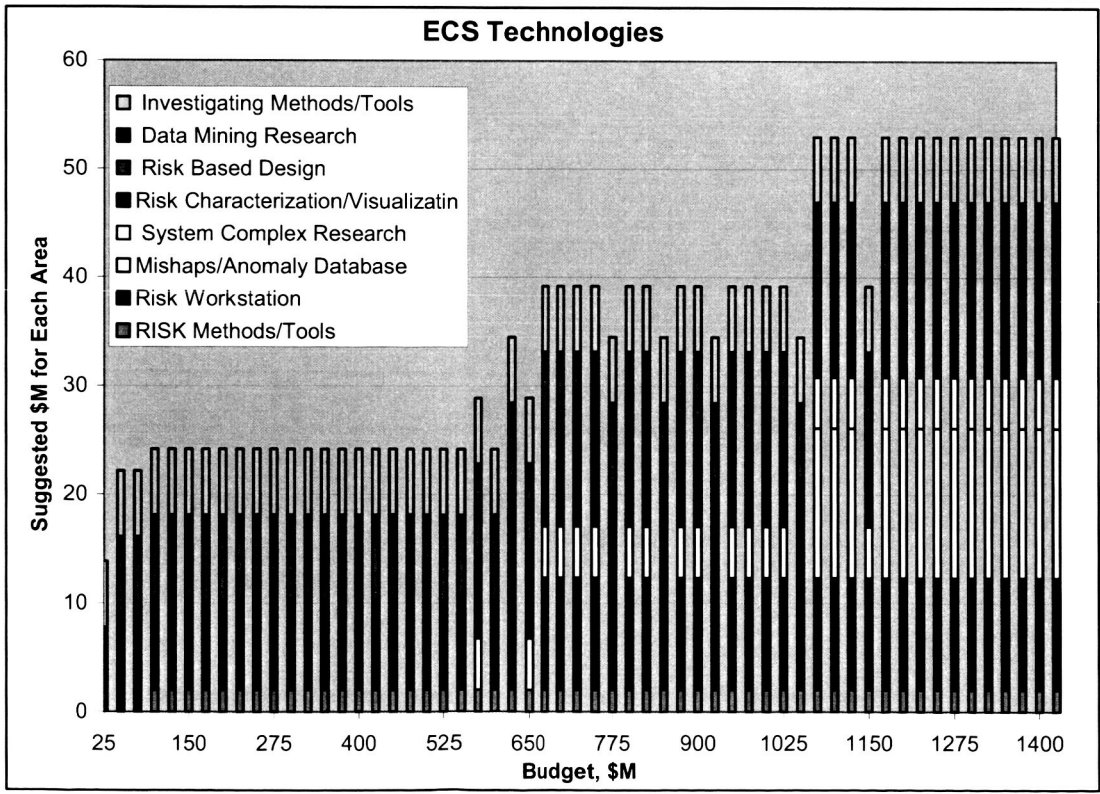
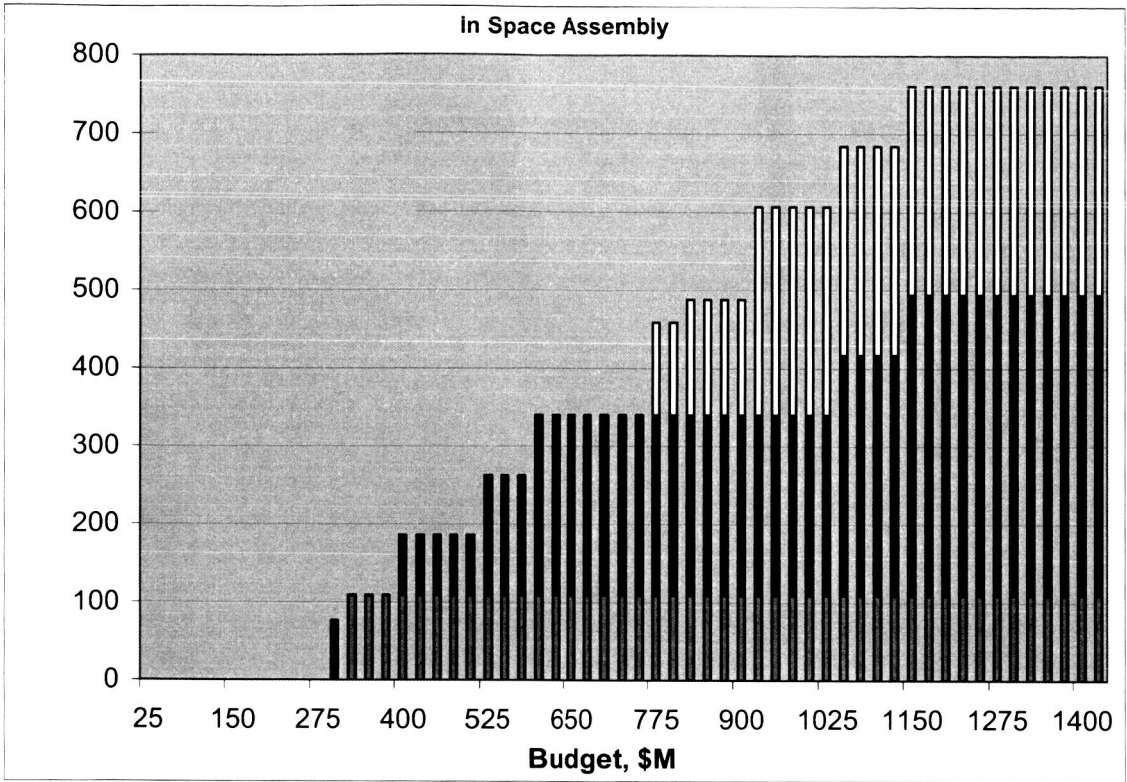
Mission-Technology Complexity Map

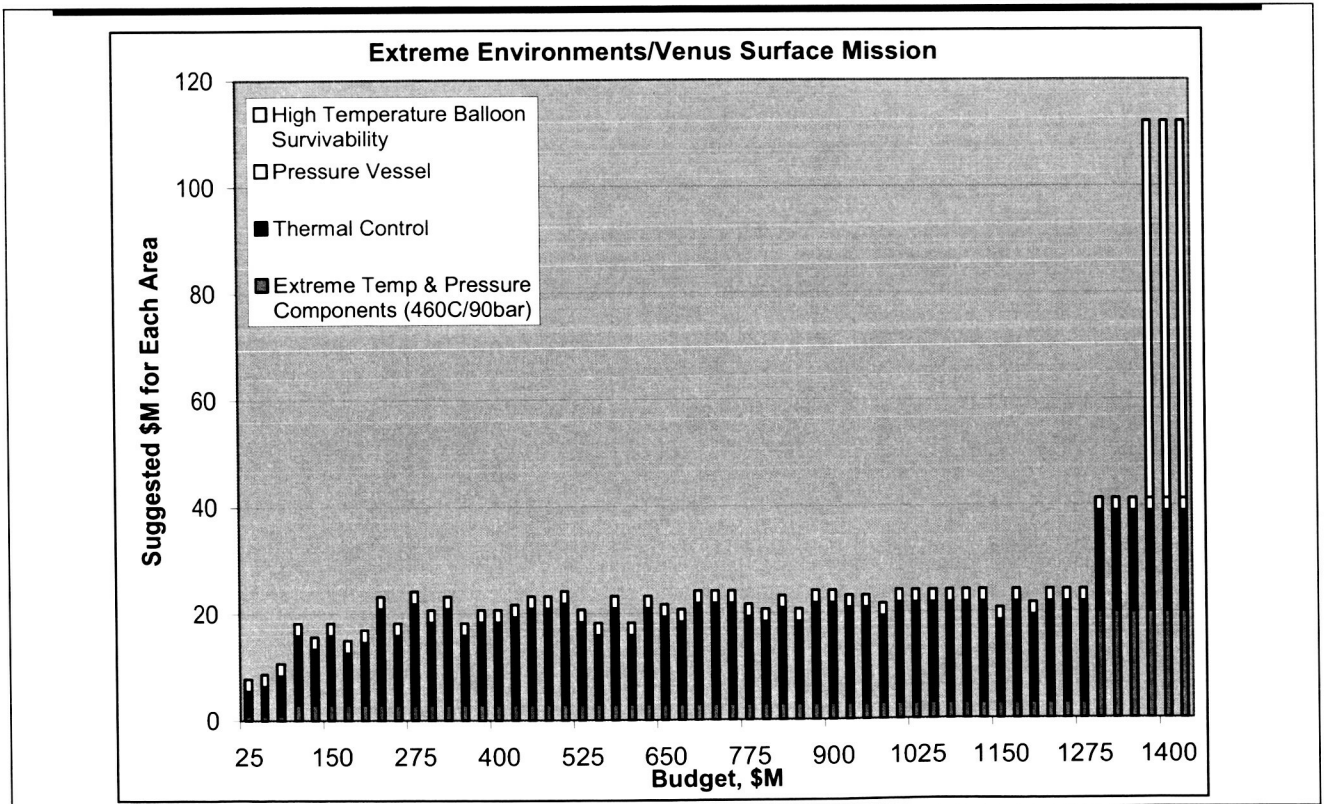
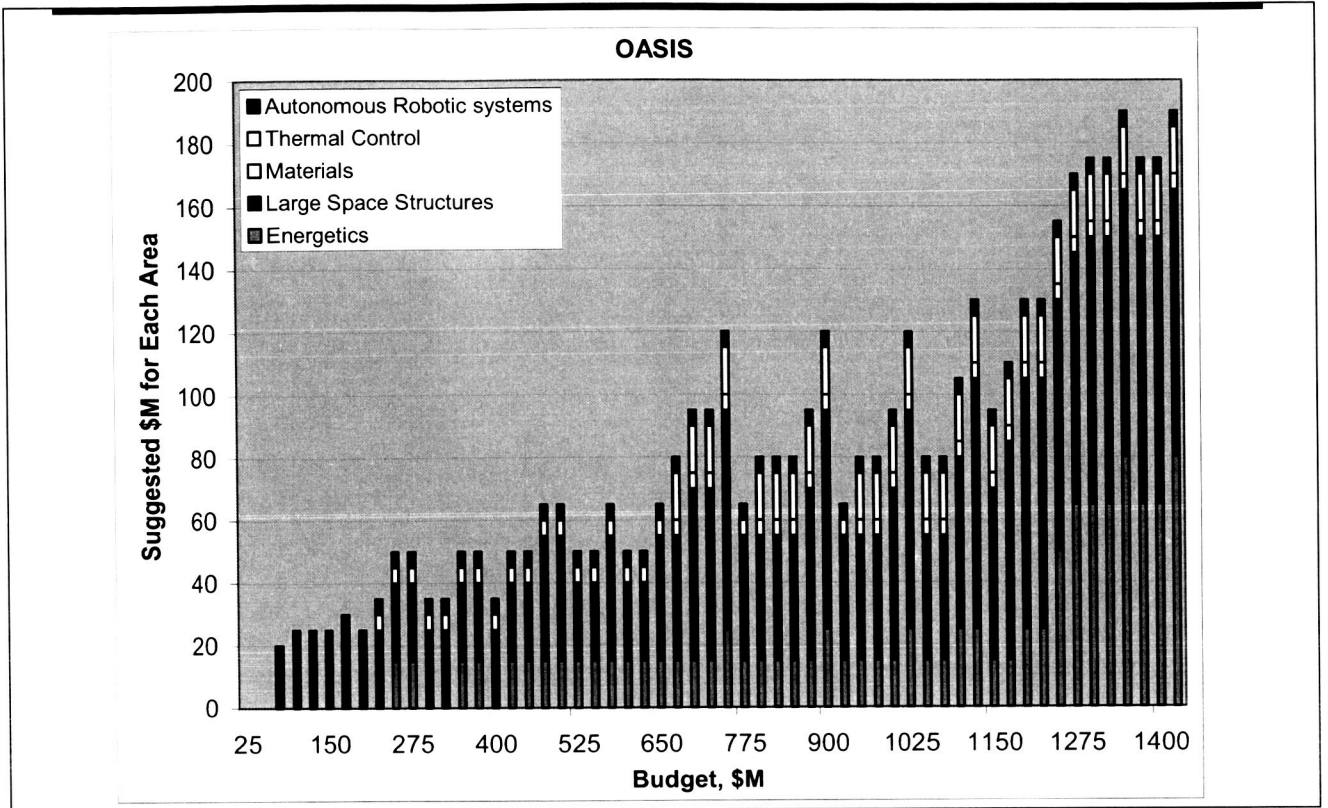


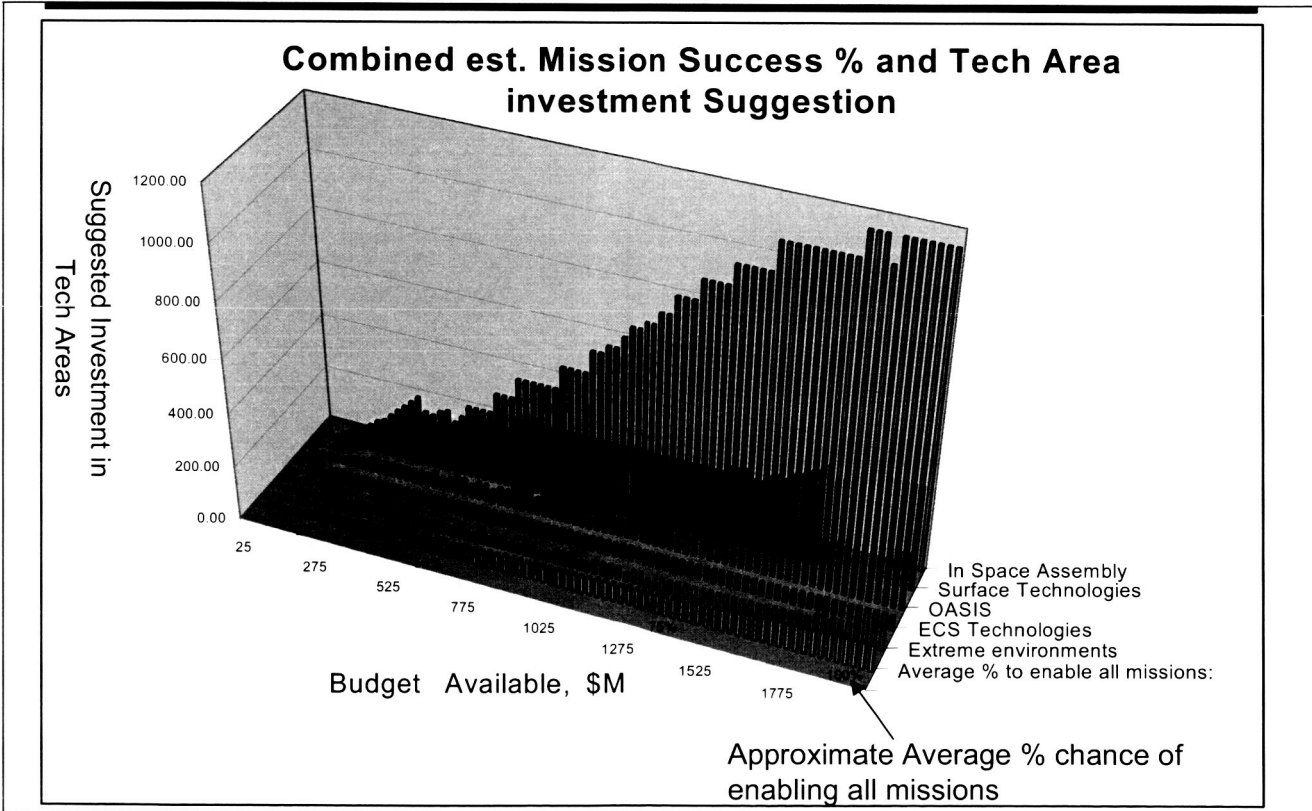
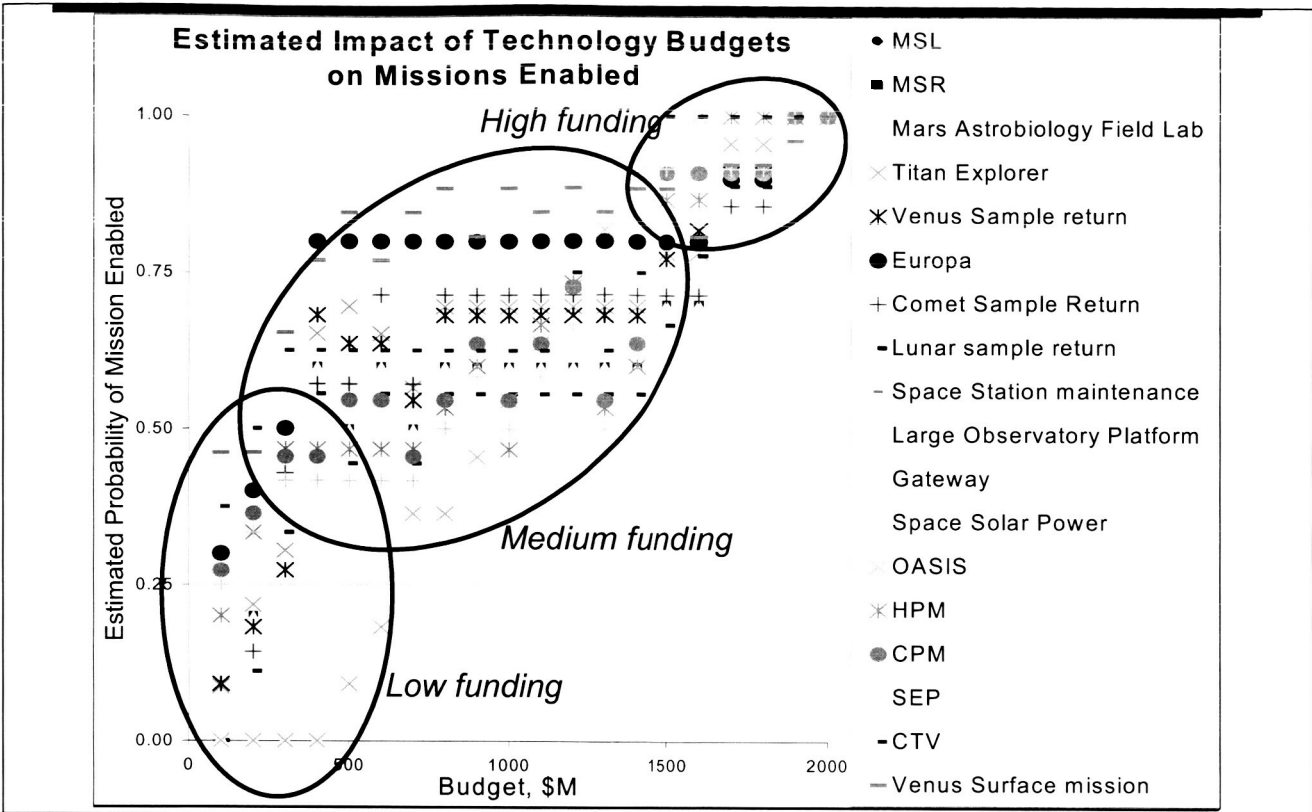
Analysis Options Used to Get Typical Results in Slides 25-30

Analysis Options Used	Other Options Available
Uniform science-return value for all missions	Can assign non-uniform science return value (user prescribed)
Uniform value for all technologies at the same hierarchical level; “democratic” hierarchy	Can prescribe general technology organizations; based for example on mission and system decomposition
Technology correlations and co-dependencies set to zero	Can explicitly include correlation & co-dependency parameters when available
Risk estimates based only on performance uncertainty	Can include cost, schedule and other risk factors
Identical development time (~10 yrs) for all technologies	Can vary technology development time as a model parameter
TRL data not included in technology projections	Can analyze TRL data within existing analysis framework









Concluding Remarks

- **Study Results to Date (January-March, 2004)**
 - Initial data base for 13 missions and 167 technology performance parameters in 23 technical areas, representing Code T,S,M,Y enterprises
 - Rapidly prototyped analysis capability to evaluate impact of technological investment on science and exploration return
- **Work Remaining (April-December, 2004)**
 - Expand data base to include more enabling missions and technologies (e.g. modular distributed structures, etc.)
 - Conduct more in-depth analysis of the representation and fidelity of the existing data set, and a more detailed treatment of the consistency and integration across program elements
 - Calibrate data base and analysis with extensive WHAT-IF computational